

COMMONWEALTH EDISON COMPANY
CALCULATION TITLE PAGE

CALCULATION NO.: L-001337

PAGE NO.: 1

☒ SAFETY RELATED

☐ REGULATORY RELATED

☐ NON-SAFETY RELATED

CALCULATION TITLE: Containment Liner Leak Chase Channel Assessment

STATION/UNIT: LaSalle County/ Unit 1

SYSTEM ABBREVIATION: NA

EQUIPMENT NO.: (IF APPL.) NA

PROJECT NO.: (IF APPL.) 10248-012

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PREPARED BY: A. C. Eberhardt/

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DATE: 10/28/97

REVISION SUMMARY: Initial Issue

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DO ANY ASSUMPTIONS IN THIS CALCULATION REQUIRE LATER VERIFICATION YES ☐ NO ☒

REVIEWED BY: R. Namperumal/

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This Calculation has been prepared using Mathcad Plus 5, S&L Program Number 03.7.548-5.0

Purpose and Objective

To perform a design assessment of 2" and 3" containment liner leak chase channels in the suppression pool and drywell. Discrepancies between design material yield strength and specified minimum material yield strength are addressed. In addition, the margin between specified yield and ultimate tensile strength as determined from vendor supplied certified material test reports (CMTRs) will be verified. Results are given in terms of margin factor which is defined as the ratio of allowable stress divided by actual stress. This calculation is performed in response to PIF # L1997-06237.

Methodology and Acceptance Criteria

This calculation shall be performed in accordance with the applicable documents listed as Design Inputs below. The acceptance criteria used for this assessment is given in Subsection 4.3.2 of the LaSalle Design Assessment Report (DAR) (D.I. 32). Applicable load combinations are defined in Table 4.3-2 of the DAR. Allowable stresses for load combinations 1 through 3 are per the 1969 AISC Specification. However, it has been demonstrated in previous calculations (see D.I. 33) that the governing load combinations involve either hydro-dynamic loads or jet impingement loads. For these load combinations, the allowable stress is limited to 0.95 fy. As permitted in Ref. 3, Sect. III.2.a, a 10% increase in minimum specified yield strength will be used to account for strain rate effects associated with impact loads. The allowable stress used for weld assessment is 21 ksi for all load combinations.

Assumptions

There are no unverified or unconservative assumptions used in this calculation.

Design Inputs (Check where applicable "APP")

APP

- 1) ☒ DC-SE-01-LS, Rev. 7, Structural Project Design Criteria
- 2) ☐ AISC's Manual of Steel Construction, Eighth Edition
- 3) ☒ AISC's Manual of Steel Construction, Seventh Edition
- 4a) ☐ ACI-318 Building Code Requirements for Reinforced Concrete Structures, 1983
- 4b) ☐ ACI-349 Building Code Requirements for NSR Concrete Structures, 1980
- 5) ☐ AISI's Cold Formed Steel Design Manual, 1983
- 6) ☒ DC-SE-02-LS, Rev. 0, Seismic Response Spectra
- 7) ☒ DC-SE-03-LS through DC-SE-09-LS, All Rev. 0, Hydrodynamic Response Spectra
- 8) ☐ Calculation 802-100.2, Rev. 0, Pages 25-37, Peak g-values for Pipe Supports
- 9) ☐ Calculation 802-100.2, Rev. 8, Pages 41-44, Peak g-values for Pipe Supports in the Drywell
- 10) ☐ Calculation 802-100.2, Rev. 8, Page 40, Peak g-values for Electrical Supports in the Drywell
- 11) ☐ Calculation 795, Rev. 0, Pages 42-43, Peak g-values for Cable Trays and Conduits
- 12) ☐ DIT No. LS-EPED-0077-01 and Calculation No. GDS-3.6.6, Rev. 1, Reduced Conduit Weights
- 13) ☐ S&L's SD&D Report No. 78, Rev. 4, Concrete Masonry Wall Design
- 14) ☐ S&L's SDS-E5.0, Rev. 1, Loads, Load Combinations, and Allowables
- 15) ☐ S&L's SDS-E11.0, Rev. 2 Hilti Kwik Bolt Design
- 16) ☐ S&L's SDS-E11.0, Rev. 3 Hilti Kwik Bolt II Design
- 17) ☐ S&L's SDS-E30.0, Rev. 0 Cable Tray Design

- 18) ☐ S&L's SDS-E31.0, Rev. 2 Cable Tray Support Design
- 19) ☐ S&L's SDS-E33.0, Rev. 3 Conduit and Conduit Support Design
- 20) ☒ S&L's SDS-E37.0, Rev. 5 Pipe Support Design
- 21) ☐ S&L's SDS-E41.0, Rev. 2 Weld Design
- 22) ☐ S&L's SDS-E44.0, Rev. 0 Embedment Plate Design
- 23) ☐ Calculation 802-104.4, Rev. 9, Pages 4-5, Design Equations for Embedment Plates
- 24) ☒ LaSalle County Station Updated Final Safety Report, Rev. 11
- 25) ☒ LaSalle County Station Drawing S-384, Rev. U
- 26) ☐ S&L's GDS-3.6.3, Rev. 3
- 27) ☐ S&L's Project Instruction LSNS-22, Rev. 3
- 28) ☒ NUREG ICR-2913, "Two Phase Jet Loads" Published Jan. 1983 by SANDIA National Laboratories
- 29) ☐ S&L's Project Instruction LSNS-16, Appendix I, Rev. 0
- 30) ☒ CBI Shop Drawings (See specific references below for Revision nos.)
- 31) ☒ SRV & LOCA Loads on Downcomer Bracing and Gusset Plates, NSLD Calc. 3C7-1179-002)
- 32) ☒ LaSalle County Station Mark II Design Assessment Report, Rev. 10
- 33) ☒ LaSalle Calculation 164, "Suppression Pool Liner Assessment & Anchorage Assessment"
- 34) ☒ LaSalle Calculation 3C7-1075-001, "Loads due to Loss-of Coolant Accident in LaSalle Containment," Rev. 6
- 35) ☒ LaSalle Specification J-2534, "Primary Containment Steel Liner Work", Amd. 3, 9-14-81

References

- 1. PIF # L1997-06237
- 2. Staad-III Structural Analysis, S&L Program # 03.7.065-20.2)
- 3. USNRC Standard Review Plan 3.6.2, Rev. 1, July 1981

GLOBAL UNIT DEFINITIONS :

$$\begin{array}{llllll} \text{kip} = 1000 \cdot \text{lbf} & \text{psf} = \frac{\text{lbf}}{\text{ft}^2} & \text{ksf} = \frac{\text{kip}}{\text{ft}^2} & \text{ksi} = \frac{\text{kip}}{\text{in}^2} & \text{ORIGIN} = 1 \\ \text{lbs} = 1 \cdot \text{lbf} & & \text{lb} = 1 \cdot \text{lbf} & & \end{array}$$

Calculations**General Description**

A total of three cases are considered to cover all variations in configuration, material properties, and governing load combinations. These three cases are described below as part of the calculation index.

Calculation Index

The calculation is divided into three sections to consider the following cases:

Pages 5 to 11, 2"x1"x 3/16" stainless steel channel subject to all loads in the suppression pool.

Pages 12 to 17, 3x4.1 channel subjected to jet impingement loads in the drywell and hydro-dynamic LOCA and SRV loads in the suppression pool.

Pages 18 to 24, 3x4.1 channel with one leg shortened by 1/2" to accommodate a thickened embed plate. This configuration is analyzed for hydro-dynamic LOCA and SRV loads in the suppression pool.

Case 1: 2"x1"x 3/16" Stainless Steel Channel**Preparation of Input to Staad Computer Model**

This model includes the 2" channel as well as the half coupling and test plug connection detail shown in Section D-D on CBI dwg. 34, Rev. 3. This model covers all 2" channels on the wall and basemat. Joint coordinates are center line coordinates. Additional joints are provided so that stresses can be determined at the end of the fillets at each corner of the channel. Member properties of the channel web and flanges are based on a thickness tw of 3/16" and an effective width of 3.0". This is based on the criteria given in D.I. 20, Fig. 37.6.6-4 where ba is the side dimension of a square with an area equivalent to that of the 1/4" half coupling which has a diameter of 7/8".

$$d = \frac{7}{8} \text{ in}$$

$$tw = \frac{3}{16} \text{ in}$$

Equivalent square: $ba = \left(\pi \cdot \frac{d^2}{4} \right)^{.5}$

$$ba = 0.78 \text{ in}$$

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STAAD MODEL FOR CASE 1 - $[2" \times 1" \times \frac{3}{16}"$
WITH TES, COUPLING

Channel Properties

$$d = 2"$$

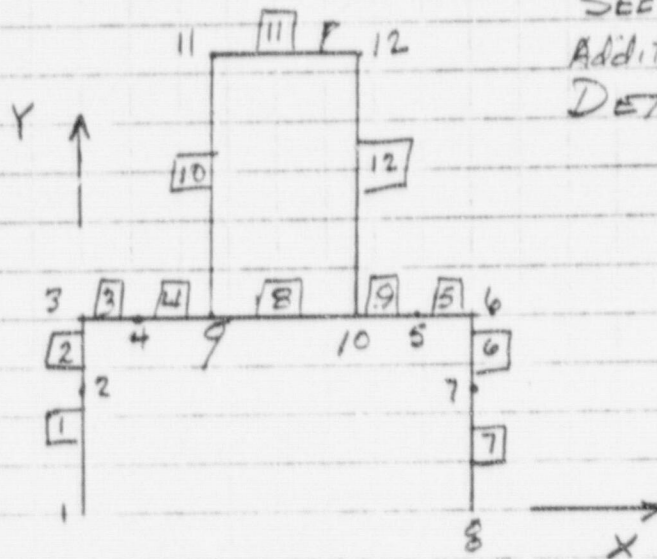
$$bf = 1"$$

$$t_f = t_w = \frac{3}{16}"$$

$$k = 0.261" \text{ conservative}$$

test couplings, see next page.

SEE NEXT 2 PAGES FOR
Additional MODEL
DETAILS



JOINT Coord.

1	0	0
2	0	.7394
3	0	.9063
4	.1669	.9063
5	.16456	.9063
6	1.8125	.9063
7	1.8125	.7394
8	1.8125	0
9	0.5475	.9063
10	1.264	.9063
11	0.5475	2.33
12	1.264	2.33

Member Incidences

1	12
2	2 3
3	3 4
4	4 9
5	5 6
6	6 7
7	7 8
8	8 9 10
9	9 10 5
10	9 11
11	11 12
12	12 10

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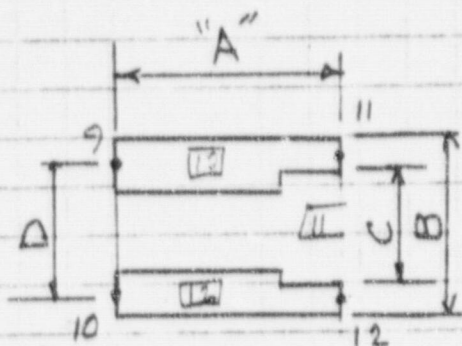
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Date

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Date

 $\frac{1}{4}" \Phi$ 3,000 lb half couplingA, B & C Dimensions From CATALOG W-6B
Bonney Forge
Allentown, PA

$$A = 1\frac{1}{8}"$$

$$B = \frac{7}{8}"$$

$$C = 0.555"$$

$$D = B - \frac{B-C}{2}$$

$$= \frac{7}{8} - \frac{\frac{7}{8} - 0.555}{2} = 0.915"$$

$$D/2 = 0.3575"$$

Dim "A" including $\frac{1}{4}"$ PLUG

$$= A + \frac{C}{2} =$$

$$1\frac{1}{8} + \frac{0.555}{2} = 1.40$$

USE 1.42"

Effective width $b = b_a + 12 \cdot t_w$

$$b = 3.0 \cdot \text{in}$$

C 2 x 1 x 3/16" Properties of Flange and Web

$$b = 3.0 \cdot \text{in}$$

$$t = \frac{3}{16} \cdot \text{in}$$

$$A_Y = b \cdot t$$

$$I_Z = \frac{b \cdot t^3}{12}$$

$$A_Y = 0.5625 \cdot \text{in}^2$$

$$I_Z = 0.00165 \cdot \text{in}^4$$

Note that A_x is calculated in the Staad program using the values of Y_D and Z_D . Arbitrary values of I_X , I_Y , and A_Z are input to Staad but are not used in the analysis.

Member Properties of 1/4" Dia. 3000 lb. half coupling

From Catalog W-6B Bonney Forge, Allentown, PA

$$A = 1.125 \cdot \text{in}$$

$$B = \frac{7}{8} \cdot \text{in}$$

$$C = .555 \cdot \text{in}$$

Wall thickness

$$t = \frac{B - C}{2}$$

$$t = 0.16 \cdot \text{in}$$

$$b = B$$

use

$$t = 0.17 \cdot \text{in}$$

$$b = 0.875 \cdot \text{in}$$

$$A_Y = b \cdot t$$

$$A_Y = 0.1488 \cdot \text{in}^2$$

$$I_Z = \frac{b \cdot t^3}{12}$$

$$I_Z = 0.0003582 \cdot \text{in}^4$$

Note that A_x is calculated in the Staad program using the values of Y_D and Z_D . Arbitrary values of I_X , I_Y , and A_Z are input to Staad but are not used in the analysis.

Loads:

The previous Calculation No. 164 (D.I. 33) identified the following critical loads: Maximum Accident Pressure = 45 psi (Ref. 32, Table 1.1-1) and LOCA Poolswell Impact.

As shown in Table 2 of D.I. 34, loads due to pool swell impact act on structures above the suppression pool located between El. 700'-2" (High water level in the suppression pool) and El. 718'-8" (Max. height of poolswell). The maximum value of pool swell impact pressure (= 643 psi) is used in this analysis.

Member Loads

Member loads due to poolswell act on members 1 and 2. The magnitude is equal to the pressure times the member width.

$$\text{Uniform load, members 1 and 2:} \quad GX = (643 + 45) \cdot 3.0 \frac{\text{lbf}}{\text{in}}$$

$$GX = 2064 \frac{\text{lbf}}{\text{in}}$$

Poolswell impact pressure on circular half coupling and plug:

From D.I. 34, Table 3, the maximum value of pool swell impact pressure acting on circular pipe sections is 48.88 psi.

$$\text{Uniform load, member 10:} \quad GX = (48.88 + 45) \cdot 0.875 \frac{\text{lbf}}{\text{in}}$$

$$GX = 82.14 \frac{\text{lbf}}{\text{in}}$$

Member loads on Members subjected to Accident Pressure = 45 psi

$$\text{Uniform load, mems. 3 to 7 & 9:} \quad GX1 = 45 \cdot 3.0 \frac{\text{lbf}}{\text{in}}$$

$$GX1 = 135.0 \frac{\text{lbf}}{\text{in}}$$

Uniform load, members 11 & 12: $GX2 = 45 \cdot 0.875 \cdot \frac{\text{lb}}{\text{in}}$

$$GX2 = 39.4 \cdot \frac{\text{lb}}{\text{in}}$$

Uniform load, member 8:

Load on member 8 is equal to balance of load not acting on member 11.

$$GY = GX1 - GX2$$

$$GY = 95.6 \cdot \frac{\text{lb}}{\text{in}}$$

Joint Loads

Member loads do not represent 100% of the pressure load acting on the channel because member length is based on center to center dimensions. Therefore, length times member width will not completely represent the total surface area subjected to pressure loads. For this reason, Joint loads are used to apply the additional load that acts on the missing surface area. Joint loads due to poolswell are applied to Joint 3. The magnitude is equal to the pressure times the missing surface area of the channel flange.

Flange length = 1"

Length modeled in Staad = .9063"

Effective width = 3.0"

Missing surface area:

$$A = (1 - .9063) \cdot \text{in} \cdot 3.0 \cdot \text{in}$$

$$A = 0.28 \cdot \text{in}^2$$

Pressure load = 688 psi

$$FX = 688 \cdot \text{psi} \cdot A$$

$$FX = 193.4 \cdot \text{lb}$$

Evaluate joint force at Joints 3 & 6 due to LOCA Pressure = 45 psi

$$FY = 45 \text{ psi} \cdot A$$

$$FY = 12.65 \cdot \text{lb}f$$

Thermal Load:

As stated in the previous Calculation No. 164 (D.I. 33) the channel is analyzed with and without the effects of temperature. The temperature change for the poolswell load condition is 76 deg. F.

Variations in Material Properties

Review of Vendor-supplied CMTRs shows that ASTM A276 Type 304 material has been supplied with two types of finishes. Hot-finished material has a yield of 30 ksi and an ultimate of 75 ksi, whereas cold-finished material has a yield of 45 ksi and an ultimate of 90 ksi. CMTRs are reviewed to determine the appropriate design yield strength for each channel configuration.

Material Yield Strength

The following documents related to Vendor Specification 2534 are provided in Attachment 4 of the calculations:

1. Stores and Metal Verification Summary & Supplement Sheets, File # 5.1 CCM
2. Material Heat Number Sheet, File No. 8.16-2, Document #18
3. Certified Material Test Reports, File No. 5.2

These documents identify the following material heats for 2x1x3/16" channels:

Heat No.	Min. Specified Yield Strength
23755	30,000 psi
F30188	30,000 psi
30660	45,000 psi
30706	45,000 psi
30844	45,000 psi
40523	30,000 psi
42379	30,000 psi

For the cases shown above, design yield strength is controlled by min. specified yield strength = 30 ksi. As permitted in Ref. 3, Section III.2.a, a 10% increase in min. specified design yield strength is used to account for strain rate effects associated with dynamic effects of LOCA loads such as jet impingement and poolswell impact loads

Therefore, the allowable stress is:

$$F_b = 0.95 \cdot 1.1 \cdot 30.0 \text{ ksi}$$

$$F_b = 31.35 \text{ ksi}$$

Results of Staad III Analysis:

Results of the analysis are shown in Attachment A.

Maximum member stress occurs in member 1 = 29.00 ksi. (See Page A7)

$$\text{Mar11} = \frac{F_b}{29.00 \text{ ksi}}$$

$$\text{Mar11} = 1.08$$

Check weld based on reactions at Joints 1 and 8 (Max. reaction occurs at joint 1).

$$\text{Weld force: } f_w = \frac{(1.434^2 + .4467^2)^{.5}}{3 \text{ in}} \cdot \text{kip}$$

$$f_w = 0.50 \cdot \frac{\text{kip}}{\text{in}}$$

$$\text{Weld Allowable: } F_w = 0.707 \cdot 21 \text{ ksi} \cdot \frac{3}{16} \text{ in}$$

$$F_w = 2.78 \cdot \frac{\text{kip}}{\text{in}}$$

$$\text{Mar12} = \frac{F_w}{f_w}$$

$$\text{Mar12} = 5.56$$

Check 3/16" weld between Channel and coupling for test plug:

Diameter of circular weld = $B = 0.875 \text{ in}$ Assume 1/2 of weld length is effective for Member no. 11 and 12, each.

Therefore, weld length = $L = 3.14159 \cdot \frac{B}{2}$

$$L = 1.37 \text{ in}$$

Weld allowable = $Fw1 = Fw \cdot L$

$$Fw1 = 3.83 \text{ kip}$$

Actual force in weld (Member 12 governs):

$$fw = (.0743^2 - .0499^2)^{.5} \cdot \text{kip}$$

$$fw = 0.09 \text{ kip}$$

$$\text{Mar13} = \frac{Fw1}{fw}$$

$$\text{Mar13} = 42.75$$

Inspection of this large margin shows that the half coupling and its attachment weld can easily resist the pool hydrodynamic loads.

Case 2: 3x4.1 Channel Subjected to Jet Impingement LoadsPreparation of Input to Staad Computer Model

This model covers the following two configurations:

1. Leak Test Assembly 39-EE located in the suppression pool (see CBI Dwg. No. 39, Rev. 5)
2. All 3" ASTM A36 leak chase channels in the drywell.

Joint coordinates are center line coordinates. Additional joints are provided so that stresses can be determined at the end of the fillets at each corner of the channel. Member properties of the channel web and flanges are based on an effective width of 1.00". Test couplings are not modeled for channels in the drywell because the design evaluation of the drywell channels is governed by jet impingement loads which are defined in D.I. #35, Sketch JF-1A on Page 2-2-7. The presence of a test coupling is not significant in the evaluation of "EE" assemblies because it is attached to a vertical channel which is not a highly loaded member in the assembly.

Member PropertiesC3x4.1 Flange Properties (Members 1, 2, 6, 7)

$$b = 1\text{-in}$$

$$t = 0.273\text{-in}$$

$$A_Y = b \cdot t$$

$$I_Z = \frac{b \cdot t^3}{12}$$

$$I_Z = 0.0016955 \cdot \text{in}^4$$

$$A_Y = 0.273 \cdot \text{in}^2$$

Note that A_x is calculated in the Staad program using the values of Y_D and Z_D . Arbitrary values of I_X , I_Y , and A_Z are input to Staad but are not used in the analysis.

C3x4.1 Web Properties (Members 3, 4, 5)

$$b = 1\text{-in}$$

$$t = 0.170\text{-in}$$

$$A = b \cdot t$$

$$I_Z = \frac{b \cdot t^3}{12}$$

$$I_Z = 0.0004094 \cdot \text{in}^4$$

$$A = 0.1700 \cdot \text{in}^2$$

Note that A_x is calculated in the Staad program using the values of Y_D and Z_D . Arbitrary values of I_X , I_Y , and A_Z are input to Staad but are not used in the analysis.

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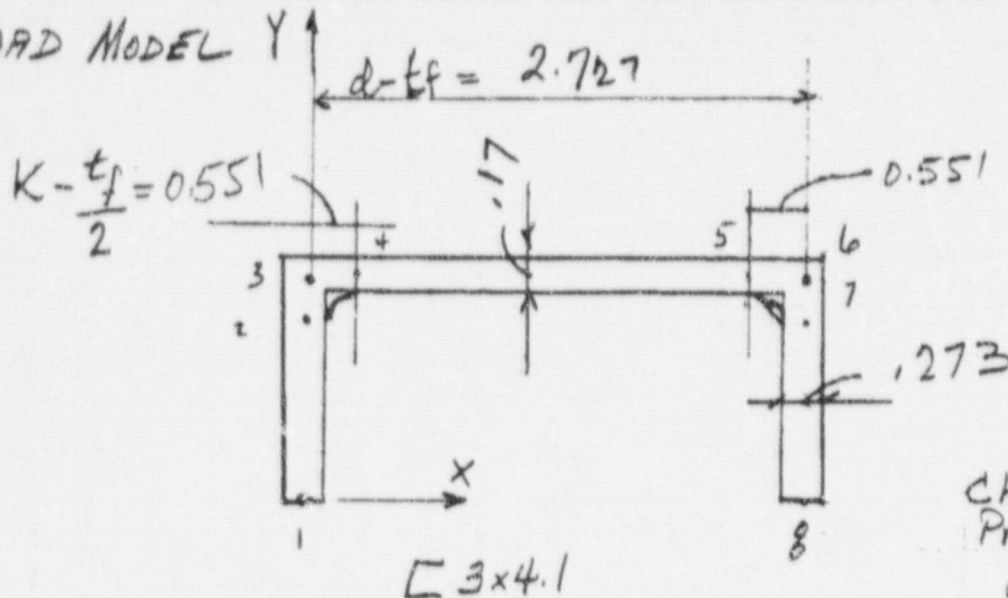
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Date

STARD MODEL



Channel Properties

$d = 3"$

$t_f = 1.41"$

$t_w = 0.273$

$k = 11/16"$

$(b_f - \frac{t_w}{2})$

JOINT COORDINATES

1	0	0
2	0	0.774
3	0	1.325
4	0.551	1.325
5	2.176	1.325
6	2.127	1.325
7	2.727	0.774
8	2.727	0

Member Incidences

1	1	2
2	2	3
3	3	4
4	4	5
5	5	6
6	6	7
7	7	8

Loads:

Critical Assembly for C3x4.1 Evaluation

Review of CBI Dwgs. 32 Rev. 4 and 39 Rev. 5 indicates that all 3" leak test channels in the suppression pool have the inside leg of the channel shortened by 1/2" to accommodate attachment to the thickened liner embed plate. See Section A-A on Dwg. 32, Section D-D on Dwg. 39 and the related table which give the "C" dimension of 1/2". The only exception to this is Leak Test Ass'y EE on Dwg. 39 which has a "C" dimension of 0". This assembly is analyzed as a separate case since its specific configuration is more severe than those channels that have a C dimension = 1/2".

The previous Calculation No. 164 (D.I. 33) identified the following critical loads: Maximum Accident Pressure = 45 psi (Ref. 32, Table 1.1-1) and LOCA Poolswell Impact.

Review of CBI Dwgs. 34 through 37, Rev. 3, 7, 3, 6, respectively, shows that Ass'y EE is used at 8 locations: Four at El. 679'-0" and four at El. 726'-0". (Dwgs 1D through 1G, Revs. 8, 7, 4, 6, respectively.) These elevations will be used to reduce the poolswell impact loads on these critical channels.

As shown in Table 2 of D.I. 34, loads due to pool swell impact act on structures above the suppression pool located between El. 700'-2" (High water level in the suppression pool) and El. 718'-8" (Max. height of poolswell).

Based on Table 2, none of the "EE" assemblies lie in the zone subject to Poolswell Impact forces.

Other forces acting on the Ass'y EE leak test channels are tabulated in the load combination table given on page 26 of Calc. 164 (Ref. 33). After pool swell impact, the next most critical load combination is #8 which is jet impingement load. This load case governs the design of A36 channels in the drywell. Jet loads in the suppression pool are much smaller than in the drywell. The jet pressure on the basemat is a maximum of 33 psi and reduces to zero at El. 700'-2" (See Fig. 4 in D.I. 34). The related joint and member loads for the jet impingement load in the drywell are described below.

After pool swell impact, the next most critical load combination for channels in the suppression pool is Case #7 which applies a vertical pressure of 197.03 psi and horz. pressure of 161.13 psi.

Results of the Ass'y EE analysis for Load Combinations 7 are presented below. Results for the other 3" channels which have a "C" dimension = 1/2" and which are located in the poolswell impact zone are evaluated later using another Staad III model (See Case 3 below).

Review of CBI Dwgs. 288 through 291 and 294 through 297, Rev. 3, 3, 3, 4, 3, 3, 2, 3, respectively, shows that 3x4.1 channels (ASTM A36) have been used as leak chase channels in the drywell. Channels in the drywell are not subject to hydrodynamic impact loads, therefore the critical load for these channels is jet impingement. See load combination #8 in the load combination table given on page 26 of Calc. 164 (Ref. 33).

Calcs. For

Calc. No. L-001327

Rev.

Date

Page 13.1 of

Safety-Related

Non-Safety-Related

Client

Project

Proj. No. 10248-C12 Equip. No.

Prepared by

Date

Reviewed by

Date

Approved by

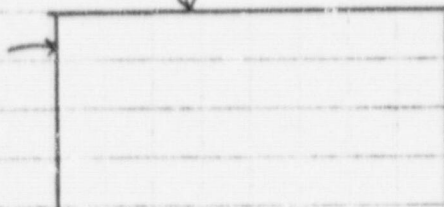
Date

STAAD ANALYSIS - LOAD CASE 1 - UNIFORM PRESSURE LOAD
 (LOAD COMBINATION #7 in CALC #164, D.I. #33)
 REVISED L.C. WHEN POOL SWELL
 DOES NOT APPLY

$$D \quad ESS \quad LOCA \quad SRV \quad Pa \\ 11.5 + 14.25 + 70.84 + 19.54 + 45 = 161.13 \text{ psi}$$

D 11.5
 ESS 4.36
 LOCA 116.56
 SRV 19.61
 Pa 45.0

197.03 psi



D 11.5
 ESS 4.36
 LOCA 45.00 } no drag load
 SRV 17.96
 Pa 45.0
 123.8 psi

Member Loads

Member loads due to Load Combination #7 applies a vertical pressure of 197.03 psi and horz. pressure of 161.13 psi. The magnitude of the member load is equal to the pressure because the member width is 1". The vertical pressure is applied to members 1 & 2 and the horz. pressure is applied to Members 3 to 5.

The load on members 6 and 7 is the same as on members 1 and 2 minus the drag loads. From Calc. 164, Page 21, the LOCA drag load is 71.56 psi and the SRV drag load is 1.67 PSI.

$$\text{Net Load on Members 6 \& 7:} \quad GX = (197.03 - 71.56 - 1.67) \cdot \text{psi}$$

$$GX = 123.80 \cdot \text{psi}$$

Jet Impingement loads on Members 3, 4, and 5

As shown on Pages 26 and 27 of Calc. 164, Load Case 8 is the jet impingement pressure load of 467 lbs/in acting on members 3, 4, and 5 which represent the channel web.

Note: See Page 25 for further discussion of jet loads in the containment drywell area.

Joint Loads

Member loads do not represent 100% of the pressure load acting on the channel because member length is based on center to center dimensions. Therefore length times member width will not completely represent the total surface area subjected to pressure loads. For this reason, Joint loads are used to apply the additional load that acts on the missing surface area. Joint loads for Load Combination 7 are applied to Joints 3 and 6. The magnitude is equal to the pressure times the missing surface area of the channel flange and web.

$$\begin{aligned} \text{Flange length} &= 1.41'' \\ \text{Length modeled in Staad} &= 1.325'' \end{aligned}$$

$$\begin{aligned} \text{Web length} &= 3'' \\ \text{Length modeled in Staad} &= 2.727 \end{aligned}$$

$$\text{Effective width} = 1.0''$$

Missing surface area:

$$Af = (1.41 - 1.325) \cdot \text{in} \cdot 1.0 \cdot \text{in}$$

$$Af = 0.0850 \cdot \text{in}^2$$

$$Aw = (3 - 2.727) \cdot 1 \cdot \text{in}^2$$

Aw is distributed to joints 3 and 6, therefore:

$$Aw = \frac{Aw}{2}$$

$$Aw = 0.14 \cdot \text{in}^2$$

Pressure load for Jt. 3 in X & Y directions = 197.03 psi & -161.13 psi, respectively

$$FX = 197.03 \text{ psi} \cdot Af$$

$$FY = -161.13 \text{ psi} \cdot Aw$$

$$FX = 16.75 \cdot \text{lb}f$$

$$FY = -21.99 \cdot \text{lb}f$$

Pressure load for Jt. 6 in X & Y directions = -123.8 psi & -161.13 psi, respectively

$$FX = -123.8 \text{ psi} \cdot Af$$

$$FY = -161.13 \text{ psi} \cdot Aw$$

$$FX = -10.523 \cdot \text{lb}f$$

$$FY = -21.99 \cdot \text{lb}f$$

Evaluate Joint Force at Joints 3 & 6 due to Jet Pressure = -467 psi

$$FY = -467 \text{ psi} \cdot Aw$$

$$FY = -63.75 \cdot \text{lb}f$$

Thermal Load:

As stated in the previous Calculation No. 164 (D.I. 33) the channel is analyzed with and without the effects of temperature. The temperature change for the poolswell load condition is 76 deg. F, and the change for Jet Impingement Load is 30 deg. F. The 30 deg. change is accomplished by using a load factor of 0.4 in Load Combination 5.

Material Yield Strength

The following documents related to Vendor Specification 2534 are provided in Attachment 4 of the calculations:

1. Stores and Metal Verification Summary & Supplement Sheets, File # 5.1 CCM
2. Material Heat Number Sheet, File No. 8.16-2, Document #18
3. Certified Material Test Reports, File No. 5.2

These documents identify the following material heats for the 3x4.1 channels in Assembly 39-EE in the suppression pool:

Heat No.	Min. Specified Yield Strength	Actual Ultimate Strength Per CMTR
40075	30,000 psi	85,350
40489	45,000 psi	89,200

For the cases shown above, design yield strength is equal to the min. specified yield strength = 30 ksi. As permitted in Ref. 3, Section III.2.a, a 10% increase in the design yield strength is used to account for strain rate effects associated with dynamic effects of LOCA loads such as jet impingement and poolswell impact loads. Therefore, the allowable stress for channels in the suppression pool is:

$$Fb1 = 0.95 \cdot 1.1 \cdot 30.0 \text{ ksi}$$

$$Fb1 = 31.35 \cdot \text{ksi}$$

The CBI CMTR documents identify the following material heats for the ASTM A36 3x4.1 channels in drywell:

Heat No.	Min. Specified Yield Strength	Actual Tensile Strength Per CMTR
93063	36,000 psi	68,100 psi
94541	36,000 psi	72,300 psi

Design yield strength is equal to the min. specified yield strength = 36 ksi. In addition, a 10% increase in min. specified design yield strength is used as permitted to account for strain rate effects associated with jet impingement loads. Therefore, the allowable bending stress for channels in the drywell is:

$$Fb2 = 0.95 \cdot 1.1 \cdot 36.0 \text{ ksi}$$

$$Fb2 = 37.62 \cdot \text{ksi}$$

Results of Staad III Analysis:

Results of the analysis are shown in Attachment B.

For the 3x4.1 channels in Assembly 39-EE in the suppression pool, the maximum member stress due to hydrodynamic load (L.C. #1) occurs in member 4 (12.57 ksi), and Fb1 is the allowable stress. For A36 channels in the drywell, the maximum member stress due to jet impingement occurs in member 4 (35.77 ksi), and Fb2 is the allowable stress.

Assembly 39-EE

ASTM A-36 Channels in Drywell

$$\text{Mar21} = \frac{\text{Fb1}}{12.57 \text{ ksi}}$$

$$\text{Mar21a} = \frac{\text{Fb2}}{35.77 \text{ ksi}}$$

$$\text{Mar21} = 2.49$$

$$\text{Mar21a} = 1.05$$

Check weld based on reactions at Joints 1 and 8 (Max. reaction occurs in L.C. #5).

$$\text{Weld force: } \text{fw} = (.2038^2 + .7005^2)^{.5} \cdot \text{kip}$$

$$\text{fw} = 0.73 \cdot \text{kip}$$

$$\text{Weld Allowable: } \text{Fw} = 0.707 \cdot 21 \cdot \text{ksi} \cdot \frac{3}{16} \cdot \text{in}^2$$

$$\text{Fw} = 2.78 \cdot \text{kip}$$

$$\text{Mar22} = \frac{\text{Fw}}{\text{fw}}$$

$$\text{Mar22} = 3.82$$

Case 3: 3x4.1 Channel With Short LegPreparation of Input to Staad Computer Model

This model covers all 3x4.1 channels billed on CBI Dwg. Nos. 32 (Rev. 4) and 39 (Rev. 5) other than those identified as Leak Test Assembly 39-EE (See Case 2). All Case 3 channels have one leg shortened by 1/2" as shown on Dwgs. 32 (Section A-A) and 39 (Section D-D).

Joint coordinates are center line coordinates. Additional joints are provided so that stresses can be determined at the end of the fillets at each corner of the channel. Member properties of the channel web and flanges are based on an effective width of 1.00". Test couplings are not modeled for this configuration because Ass'y A (Dwg. 32) and Detail E (Dwg. 39) show that test couplings on these assemblies are always mounted on a vertical leg of the assembly. The vertical leg is not the critical member for pool hydrodynamic loads because these impact loads primarily act in the vertical direction. Thus, the presence of a test coupling in the suppression pool is only significant when it is attached to a horizontal channel.

Member Properties

Member properties are the same as those used in the Case 2 model.

Material Yield Strength

The following documents related to Vendor Specification 2534 are provided in Attachment 4 of the calculations:

1. Stores and Metal Verification Summary & Supplement Sheets, File # 5.1 CCM
2. Material Heat Number Sheet, File No. 8.16-2, Document #18
3. Certified Material Test Reports, File No. 5.2

These documents identify the following material heats for the 3x4.1 channels billed on CBI Dwgs. 32 and 39:

Heat No.	Min. Specified Yield Strength	Actual Ultimate Strength per CMTR
30284	45,000 psi	91,400
40079	30,000 psi	85,350
40489	45,000 psi	89,200

Calc. For STARD MODEL - CASE 3Calc. No. L-001337Rev. 0 DatePage 18.1 of

Safety-Related

Non-Safety-Related

Client

Project

Proj. No. 1024B-012 Equip. No.

Prepared by

Date

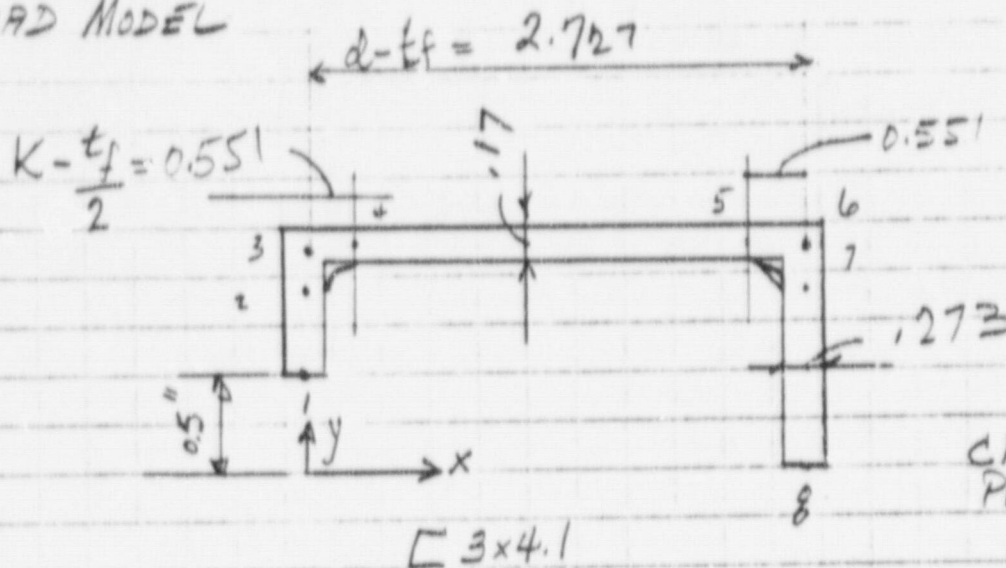
Reviewed by

Date

Approved by

Date

STARD MODEL



Channel Properties

 $d = 3"$ $bf = 1.41"$ $tf = 0.273$ $tw = 0.17$ $k = 1/16"$

JOINT COORDINATES

1	0	0.5
2	0	0.774
3	0	1.325
4	0.551	1.325
5	2.176	1.325
6	2.727	1.325
7	2.727	0.774
8	2.727	0

 $(bf - tw)$

Member Incidences

1	1	2
2	2	3
3	3	4
4	4	5
5	5	6
6	6	7
7	7	8

Heat number 40079 has the lowest yield strength ($F_y = 30$ ksi, so the allowable stress is:

$$F_{b1} = 0.95 \cdot 1.1 \cdot 30.0 \text{ ksi}$$

$$F_{b1} = 31.35 \text{ ksi}$$

Loads:

Critical Assembly for C3x4.1 Evaluation:

Review of the above specified CMTRs shows that channels from heat no. 40079 have a yield strength of 30 ksi. Based on previous calculations, it is necessary to reexamine the poolswell impact loads acting on the channels from heat no. 40079.

As shown in Table 2 of D.I. 34, loads due to poolswell impact act on structures above the suppression pool located between El. 700'-2" (High water level in the suppression pool) and El. 718'-8" (Max. height of poolswell). Therefore, CBI dwgs. 34 through 37 (Revision Nos. 3, 7, 4, 6, respectively) were reviewed to identify all '32'- and '39'-Series channel assemblies that fall within the poolswell zone. The following four assemblies were found to be subjected to significant poolswell impact loads: 39-A, 39-M, 39-N, 39-FF. Of these four, only Assembly 39-M had horizontal channels from heat no. 40079.

The center line elevation of 39-M is 703'-6" (See CBI dwgs. 1E and 1F, Revs. 7 & 4, respectively). Due to the poolswell phenomenon, the most critical load condition occurs when the lower flange of the bottom channel is impacted due to poolswell. This elevation corresponds to 702'-11 1/4". However, conservatively, poolswell for this case will be based on El. 704'-0". From Table 2 of D.I. 34, the poolswell impact at 704'-0" is 421.9 psi (A value of 422 psi will be used in this calculation. LOCA pressure is added to yield a total applied pressure of $422 + 45 = 467$ psi.

Other forces acting on Ass'y 39-M are tabulated in the load combination table given on page 26 of Calc. 164 (Ref. 33). As stated earlier, jet impingement load is not a governing load in the suppression pool. Therefore, after pool swell impact, the next most critical load combination for channels in the suppression pool is Case #7 which applies a vertical pressure of 197.03 psi and horz. pressure of 161.13 psi.

The Staad model for this third configuration is analyzed for a total of nine load cases as described below:

Case 1. This case represents poolswell impact on the lower flange of the upper channel of Ass'y 39-FF.

Case 2. This case represents poolswell impact on the lower flange of the lower channel of Ass'y 39-FF. This case combined with thermal load produces the highest stresses and is used to qualify all channel assemblies built using heat numbers other than 40079.

Case 3. This case represents poolswell impact on the lower flange of the lower channel of Ass'y 39-M.

Case 4. This case represents Case #7 which applies a vertical pressure of 197.03 psi and horz. pressure of 161.13 psi.

Case 5. This case represents thermal effects due to accident temperature.

Cases 6, 7, 8, and 9 are the same as cases 1, 2, 3, and 4 combined with the appropriate thermal effects.

Member Loads

Load Case 1

Member loads due to poolswell + accident pressure acting on members 1 and 2. The magnitude is equal to the pressure times the member width.

$$\text{Uniform load, Members 1 and 2:} \quad GX = (643 + 45) \cdot 1.0 \cdot \frac{\text{lb}}{\text{in}}$$

$$GX = 688 \cdot \frac{\text{lb}}{\text{in}}$$

Member loads on Members subjected to Accident Pressure = 45 psi

$$\text{Uniform load, Members 3 to 7:} \quad GX1 = 45 \cdot 1.00 \cdot \frac{\text{lb}}{\text{in}}$$

$$GX1 = 45.00 \cdot \frac{\text{lb}}{\text{in}}$$

Joint Loads

Joint loads are applied to Joints 3 and 6 to account for the additional poolswell load or accident pressure that acts on surface area not represented by member length times width. The magnitude is equal to the pressure times the missing surface area of the channel flange and web.

Flange length = 1.41"
Length modeled in Staad = 1.325"

Web length = 3"
Length modeled in Staad = 2.727

Effective width = 1.0"

Missing surface area:

$$A_f = (1.41 - 1.325) \cdot \text{in} \cdot 1.0 \cdot \text{in}$$

$$A_f = 0.0850 \cdot \text{in}^2$$

$$A_w = (3 - 2.727) \cdot 1 \cdot \text{in}^2$$

A_w is distributed to joints 3 and 6, therefore:

$$A_w = \frac{A_w}{2}$$

$$A_w = 0.14 \cdot \text{in}^2$$

Pressure load for Jt. 3 in X & Y directions = 688 psi & -45 psi, respectively

$$F_X = 688 \cdot \text{psi} \cdot A_f$$

$$F_X = 58.48 \cdot \text{lb}_f$$

$$F_Y = -45 \cdot \text{psi} \cdot A_w$$

$$F_Y = -6.14 \cdot \text{lb}_f$$

Pressure load for Jt. 6 in X & Y directions = -45 psi

$$F_X = -45 \cdot \text{psi} \cdot A_f$$

$$F_X = -3.825 \cdot \text{lb}_f$$

$$F_Y = -45 \cdot \text{psi} \cdot A_w$$

$$F_Y = -6.14 \cdot \text{lb}_f$$

Load Case 2

Load Case 2 is the same as Load case 1 except that poolswell impact pressure acts on Members 6 and 7 and Joint 6 instead of members 1 and 2 and Joint 3. Therefore, the loads for these members and joints are just the reverse case of Load Case 1.

Load Case 3

Load Case 3 represents poolswell impact on the lower flange of the lower channel of Ass'y 39-M. Member loads due to poolswell + accident pressure act on members 6 and 7. The magnitude is equal to the pressure times the member width. As stated earlier, the center line elevation of 39-M is 703'-6" (See CBI dwgs. 1E and 1F, Revs. 7 & 4, respectively). Due to the poolswell phenomenon, the most critical load condition occurs when the lower flange of the bottom channel is impacted due to poolswell. This elevation corresponds to 702'-11 1/4". However, conservatively, poolswell for this case will be based on El. 704'-0". From Table 2 of D.I. 34, the poolswell impact at 704'-0" is 421.9 psi (A value of 422 psi will be used in this calculation. LOCA pressure is added to yield a total applied pressure of $422 + 45 = 467$ psi).

Uniform load, Members 6 and 7:

$$GX = (422 + 45) \cdot 1.0 \cdot \frac{\text{lb}}{\text{in}}$$
$$GX = 467 \cdot \frac{\text{lb}}{\text{in}}$$

Member loads on members subjected to Accident Pressure = 45 psi

Uniform load, Members 1 to 5:

$$GX1 = 45 \cdot 1.00 \cdot \frac{\text{lb}}{\text{in}}$$
$$GX1 = 45.00 \cdot \frac{\text{lb}}{\text{in}}$$

Joint Loads:

Joint loads are the same as for load case 2 except for Joint 6. Pressure load for Jt. 3 in the -X direction = 467 psi:

$$FX = 467 \cdot \text{psi} \cdot Af$$

$$FX = 39.69 \cdot \text{lb}$$

Load Case 4: Alternate Load Case 7 (Calculation 164)

As shown on Pages 26 of Calc. 164, there is an alternate Load Combination #7 which applies a vertical pressure of 197.03 psi and horz. pressure of 161.13 psi.. The magnitudes of the member loads and the joint loads are the same as Case 2 given on page 14 of this calculation except that the loads for these members and joints are just the reverse case, i.e., the maximum pressure load (197.03 psi) acts on Members 6 and 7 and Joint 6 instead of Members 1 and 2 and Joint 3.

Load Case 5: Thermal Load

As stated in the previous Calculation No. 164 (D.I. 33) the channel is analyzed with and without the effects of temperature. Since jet impingement load in the suppression pool is not a governing load condition, the temperature change corresponding to the poolswell load condition of 76 deg. F is used in the analysis.

Loads Cases 6, 7, 8 and 9:

These cases are the same as Cases 1, 2, 3, and 4 combined with thermal effects.

Results of Staad III Analysis:

Results of the analysis are shown in Attachment C. Review of member stresses shown on Page C7 shows that design of channels made from Heat No. 40079 is governed by Load Combination 8 ($f_8 = 26.17$ ksi), and design of all other 3x4.1 channels covered in Case 3 is governed by Load Combination 7 ($f_7 = 39.04$ ksi).

$$f_8 = 26.17 \text{ ksi}$$

$$f_7 = 39.04 \text{ ksi}$$

Procedure for Adjusting Design Yield Stress

The CMTR for heat no. 40489 reports a yield of 51.6 ksi and an ultimate of 89.2 ksi. The yield meets the ASTM specified minimum of 45 ksi for cold-finished material, but the ultimate value is 0.8 ksi less than the minimum value of 90 ksi. In order to maintain the ASTM specified stress margin of 45 ksi between yield and ultimate, the design yield stress is adjusted to meet the following criteria:

$$F_y \max \leq F_u - 45 \text{ ksi.}$$

Allowable Stress:

As stated previously, the 3x4.1 channel covered under this Case 3 analysis were manufactured from the following three heat numbers:

Heat No.	Min. Specified Yield Strength	Ultimate Strength from CMTR	
30384	45,000 psi	91,400	> 90,000 O.K.
40079	30,000 psi	85,350	> 75,000 O.K.
40489	45,000 psi	89,200	Revised $F_y = 89,200 - 45,000 = 44,200$

The heat number corresponding to Load Cases 3, 4, 8, and 9 is 40079, so $F_{y8} = 30.0$ ksi.

Due to the adjusted design yield strength, Heat No. 40489 has the next lowest effective yield strength and controls for Load Case 7, so $F_{y7} = 44.2$ ksi

The corresponding allowable stresses are:

$$Fb8 = 0.95 \cdot 1.1 \cdot 30.0 \text{ ksi}$$

$$Fb7 = 0.95 \cdot 1.1 \cdot 44.2 \text{ ksi}$$

$$Fb8 = 31.35 \text{ ksi}$$

$$Fb7 = 46.19 \text{ ksi}$$

The resulting design margins are:

$$Mar31 = \frac{Fb8}{f8}$$

$$Mar31a = \frac{Fb7}{f7}$$

$$Mar31 = 1.20$$

$$Mar31a = 1.18$$

Check weld based on reactions at Joints 1 and 8 (Max. reaction occurs in L.C. #2 at Jt. 8):

$$\text{Weld force: } fw = (.5924^2 + .1064^2)^{.5} \cdot \text{kip}$$

$$fw = 0.60 \text{ kip}$$

$$\text{Weld Allowable: } Fw = 0.707 \cdot 21 \text{ ksi} \cdot \frac{3}{16} \text{ in}^2$$

$$Fw = 2.78 \text{ kip}$$

$$Mar32 = \frac{Fw}{fw}$$

$$Mar32 = 4.63$$

Discussion of Jet Impingement Loads on Liner Leak Chase Channels Inside Containment Drywell

In the previous calculation (D.I. 33), the main purpose of the analysis was to analyze channels subject to hydrodynamic impact loads in the suppression pool. The jet impingement forces were not part of the governing load case in the suppression pool. This calculation includes qualification of ASTM A36 channels in the drywell, so a study is included here to document that jet loads act on the web of the leak chase channels and do not apply significant loading to the channel flanges.

The UFSAR (D.I. 24), Section 3.6, Table 3.6-8 provides a list of postulated pipe breaks inside the containment. Two types of breaks are postulated: Circumferential and Longitudinal. Circumferential breaks cause jets to develop parallel to the axis to the pipe, whereas Longitudinal breaks cause jets which act perpendicular to the axis of the pipe. Review of Figs. 3.6-1 through 3.6-15 shows that large bore high pressure piping inside the containment is close to the liner only at the containment wall penetration locations and these are always horizontal pipe runs. Vertical pipe runs are always away from the containment liner plate. Therefore, a jet force would apply a significant force to the channel flanges only if there is a postulated longitudinal pipe break at or very close to the containment wall pipe penetration.

Figs. 3.6-1 through 3.6-15 and Table 3.6-8 can be used to determine the approximate location of all longitudinal breaks inside the containment. Based on this information, the closest longitudinal breaks to the liner channels are the break numbers C214 and C214S shown in Figs. 3.6-3 and 3.6-6, respectively. Per Table 3.6-8, the pipewhip restraints which have been designed to restrain the pipe from whipping due to the postulated breaks at C214 and C214S are R-72 and R-88, respectively. These restraints are shown on LaSalle Dwg S-384, Rev. U (D.I. 25). Using the dimensions shown on the plan view, the radial distance from the RPV center line to these restraints is:

$$D1 = \left[(27\text{ ft} + 7.125\text{ in})^2 - (3\text{ ft} + 8\text{ in})^2 \right]^{.5}$$

$$D1 = 27.84\text{ ft}$$

The elbow at which the break occurs is located further from the liner than the restraint. Therefore, the location of the restraint conservatively can be used as the break location. The sketch on the following page shows that the distance from the pipe break to the drywell liner is greater than 19.5 ft. According to NUREG \CR-2913 (D.I. 28), pipe break forces are not significant at distances greater than 8 pipe diameters from the pipe break location. For the Main Steam line with a diameter of 26 in, this distance is:

$$DIA = 26\text{ in}$$

$$D8 = 8 \cdot DIA$$

$$D8 = 17.33\text{ ft}$$

The distance to the liner is 19.5 ft which is greater than D8. Therefore, it can be concluded that none of the longitudinal or circumferential breaks inside the containment will apply a significant jet loads on the flanges of the ASTM A36 liner leak chase channels in the containment drywell.

Client

Project

Proj. No. 10248-12

Equip. No.

Prepared by

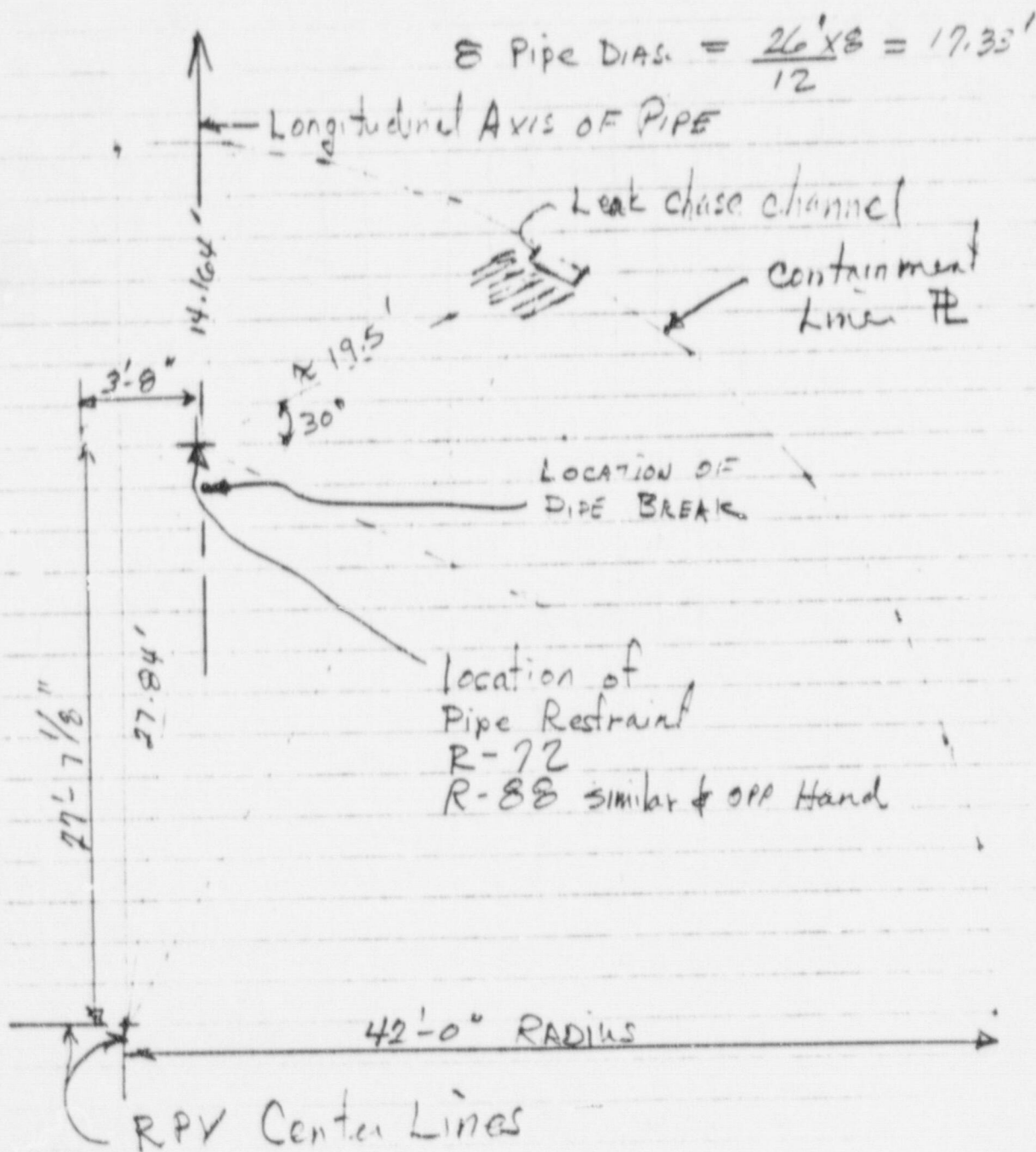
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Reviewed by

Date

Approved by

Date



SUMMARY AND CONCLUSIONS

In response to PIF # L1997-06237, three channel configurations have been evaluated in the suppression pool and the drywell. This assessment has been performed using minimum specified material yield strength as determined from vendor supplied certified material test reports (CMTRs). In addition, the design yield strength is reduced, if necessary, to maintain a minimum of 45 ksi between yield and ultimate. Results are given in terms of the margin factor which is defined as the ratio of allowable stress divided by actual stress.

All margin factors are greater than 1.0 thus demonstrating that all leak chase channels have been found to meet the acceptance criteria and are thus able to serve as part on the containment pressure boundary.

Margin factors are summarized below.

	<u>Channel Section</u>	<u>Weld</u>
Case 1:		
C 2 x 1 x 3/16	Mar11 = 1.08	Mar12 = 5.56
Case 2:		
Assembly 39-EE	Mar21 = 2.49	Mar22 = 3.82
Drywell Channels	Mar21a = 1.05	
Case 3:		
C3x4.1, Heat No. 40079	Mar31 = 1.20	Mar32 = 4.63
All other C3x4.1 in Suppression Pool	Mar31a = 1.18	


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*          S T A A D - III
*          Revision 20.2
*          Proprietary Program of
*          Research Engineers, Inc.
*          Date=   OCT 22, 1997
*          Time=   9: 5:13
*
*          USER ID: SARGENT & LUNDY ENGRS
*****

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1. STAAD PLANE 2X1X3/16 CHANNEL & TEST PLUG
2. INPUT WIDTH 72
3. UNIT INCHES POUND
4. JOINT COORDINATES
5. 1 0. 0.; 2 0. .7394; 3 0. .9063; 4 .1669 .9063; 5 1.6456 .9063
6. 6 1.8125 .9063; 7 1.8125 .7394; 8 1.8125 0.
7. 9 0.5475 .9093; 10 1.264 .9063; 11 .5475 2.33; 12 1.264 2.33
8. MEMBER INCIDENCES
9. 1 1 2; 2 2 3; 3 3 4; 4 4 9; 5 5 6; 6 6 7; 7 7 8
10. 8 9 10; 9 10 5; 10 9 11; 11 11 12; 12 10 12
11. MEMBER PROPERTY AMERICAN
12. 1 TO 9 PRI IX .1 IY 8.E-4 IZ 1.65E-3 AY .5625 AZ .3 YD .1875 ZD 3.00
13. 10 11 12 PRI IX .1 IY 4E-4 IZ 3.582E-4 AY .1488 AZ .1 YD .17 ZD .875
14. MEMBER RELEASE
15. 10 12 START MZ
16. CONSTANTS
17. E STEEL ALL
18. POISSON STEEL ALL
19. DENSITY STEEL ALL
20. BETA 0. ALL
21. * STAINLESS STEEL
22. ALPHA 99E-7 ALL
23. SUPPORTS
24. 1 8 PINNED
25. LOAD 1 LC 4,5,7
26. JOINT LOAD
27. 3 FX 193.4 FY -12.65
28. 6 FX -12.65 FY -12.65
29. MEMBER LOAD
30. 1 2 UNI GX 2064.
31. 3 4 5 9 UNI GY -135.
32. 6 7 UNI GX -135.
33. 8 UNI GY -95.6
34. 10 UNI GX 82.14
35. 11 UNI GY -39.4
36. 12 UNI GX -39.4
37. LOAD 2 THERMAL
38. TEMP LOAD
39. 3 4 5 8 9 11 TEMP 76.
40. LOAD COMB 3
41. 1 1. 2 1.

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 2

42. LOAD LIST ALL
43. PERFORM ANALYSIS

PROBLEM STATISTICS

NUMBER OF JOINTS/MEMBER+ELEMENTS/SUPPORTS = 12/ 12/ 2
ORIGINAL/FINAL BAND-WIDTH = 5/ 2
TOTAL PRIMARY LOAD CASES = 2, TOTAL DEGREES OF FREEDOM = 32
SIZE OF STIFFNESS MATRIX = 288 DOUBLE PREC. WORDS
REQRD/AVAIL. DISK SPACE = 12.02/ 1069.5 MB, EXMEM = 1.02 MB

++ PROCESSING ELEMENT STIFFNESS MATRIX.	9: 5:14
++ PROCESSING GLOBAL STIFFNESS MATRIX.	9: 5:14
++ PROCESSING TRIANGULAR FACTORIZATION.	9: 5:14
++ CALCULATING JOINT DISPLACEMENTS.	9: 5:14
++ CALCULATING MEMBER FORCES.	9: 5:14

44. PRINT JOINT DISP LIST 3 6

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 3

JOINT DISPLACEMENT (INCH RADIANS) STRUCTURE TYPE = PLANE

JOINT	LOAD	X-TRANS	Y-TRANS	Z-TRANS	X-ROTAN	Y-ROTAN	Z-ROTAN
3	1	0.00652	0.00002	0.00000	0.00000	0.00000	-0.00293
	2	-0.00068	0.00000	0.00000	0.00000	0.00000	0.00056
	3	0.00585	0.00002	0.00000	0.00000	0.00000	-0.00237
6	1	0.00645	-0.00004	0.00000	0.00000	0.00000	-0.00358
	2	0.00068	0.00000	0.00000	0.00000	0.00000	-0.00056
	3	0.00713	-0.00004	0.00000	0.00000	0.00000	-0.00414

***** END OF LATEST ANALYSIS RESULT *****

45. PRINT MEMBER FORCE LIST 1 4 7 9 10 12

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 4

MEMBER END FORCES STRUCTURE TYPE = PLANE

ALL UNITS ARE -- POUN INCH

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
1	1	1	-446.67	1434.44	0.00	0.00	0.00	0.00
		2	446.67	91.68	0.00	0.00	0.00	496.42
	2	1	0.00	-32.51	0.00	0.00	0.00	0.00
		2	0.00	32.51	0.00	0.00	0.00	-24.04
	3	1	-446.67	1401.93	0.00	0.00	0.00	0.00
		2	446.67	124.19	0.00	0.00	0.00	472.38
4	1	4	625.78	-486.80	0.00	0.00	0.00	-373.83
		9	-625.38	538.18	0.00	0.00	0.00	178.77
	2	4	32.51	-0.26	0.00	0.00	0.00	29.46
		9	-32.51	0.26	0.00	0.00	0.00	-29.56
	3	4	658.29	-487.06	0.00	0.00	0.00	-344.36
		9	-657.89	538.44	0.00	0.00	0.00	149.21
7	1	7	716.66	654.98	0.00	0.00	0.00	447.39
		8	-716.66	-555.17	0.00	0.00	0.00	0.00
	2	7	0.00	32.51	0.00	0.00	0.00	24.04
		8	0.00	-32.51	0.00	0.00	0.00	0.00
	3	7	716.66	687.50	0.00	0.00	0.00	471.43
		8	-716.66	-587.68	0.00	0.00	0.00	0.00
9	1	10	690.17	-629.97	0.00	0.00	0.00	192.75
		5	-690.17	681.48	0.00	0.00	0.00	-442.97
	2	10	32.51	0.00	0.00	0.00	0.00	29.46
		5	-32.51	0.00	0.00	0.00	0.00	-29.46
	3	10	722.68	-629.97	0.00	0.00	0.00	222.21
		5	-722.68	681.48	0.00	0.00	0.00	-472.43
10	1	9	-46.06	66.85	0.00	0.00	0.00	0.00
		11	46.06	49.85	0.00	0.00	0.00	12.08
	2	9	0.00	0.00	0.00	0.00	0.00	0.00
		11	0.00	0.00	0.00	0.00	0.00	-0.01
	3	9	-46.06	66.84	0.00	0.00	0.00	0.00
		11	46.06	49.85	0.00	0.00	0.00	12.07
12	1	10	74.29	-6.25	0.00	0.00	0.00	0.00
		12	-74.29	-49.85	0.00	0.00	0.00	31.04
	2	10	0.00	0.00	0.00	0.00	0.00	0.00
		12	0.00	0.00	0.00	0.00	0.00	0.01
	3	10	74.29	-6.24	0.00	0.00	0.00	0.00
		12	-74.29	-49.85	0.00	0.00	0.00	31.05

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 5

***** END OF LATEST ANALYSIS RESULT *****

46. PRINT SUPPORT REACTIONS

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 6

SUPPORT REACTIONS -UNIT POUN INCH STRUCTURE TYPE = PLANE

JOINT	LOAD	FORCE-X	FORCE-Y	FORCE-Z	MOM-X	MOM-Y	MOM Z
1	1	-1434.44	-446.67	0.00	0.00	0.00	0.00
	2	32.51	0.00	0.00	0.00	0.00	0.00
	3	-1401.93	-446.67	0.00	0.00	0.00	0.00
8	1	-555.17	716.66	0.00	0.00	0.00	0.00
	2	-32.51	0.00	0.00	0.00	0.00	0.00
	3	-587.68	716.66	0.00	0.00	0.00	0.00

***** END OF LATEST ANALYSIS RESULT *****

47. PRINT MEMBER STRESSES LIST 1 4 7 9

2X1X3/16 CHANNEL & TEST PLUG

-- PAGE NO. 7/FINAL

MEMBER STRESSES

ALL UNITS ARE POUN/SQ INCH

MEMB	LD	SECT	AXIAL	BEND-Y	BEND-Z	COMBINED	SHEAR-Y	SHEAR-Z
1	1	.0	794.1 T	0.0	0.0	794.1	2550.1	0.0
		1.00	794.1 T	0.0	28205.6	28999.7	163.0	0.0
	2	.0	0.0 T	0.0	0.0	0.0	57.8	0.0
		1.00	0.0 T	0.0	1365.8	1365.8	57.8	0.0
	3	.0	794.1 T	0.0	0.0	794.1	2492.3	0.0
		1.00	794.1 T	0.0	26839.8	27633.9	220.8	0.0
4	1	.0	1112.5 C	0.0	21240.2	22352.8	865.4	0.0
		1.00	1111.8 C	0.0	10157.3	11269.0	956.8	0.0
	2	.0	57.8 C	0.0	1674.1	1731.9	0.5	0.0
		1.00	57.8 C	0.0	1679.6	1737.4	0.5	0.0
	3	.0	1170.3 C	0.0	19566.2	20736.4	865.9	0.0
		1.00	1169.6 C	0.0	8477.6	9647.2	957.2	0.0
7	1	.0	1274.1 C	0.0	25420.0	26694.1	1164.4	0.0
		1.00	1274.1 C	0.0	0.0	1274.1	987.0	0.0
	2	.0	0.0 C	0.0	1365.8	1365.8	57.8	0.0
		1.00	0.0 C	0.0	0.0	0.0	57.8	0.0
	3	.0	1274.1 C	0.0	26785.8	28059.9	1222.2	0.0
		1.00	1274.1 C	0.0	0.0	1274.1	1044.8	0.0
9	1	.0	1227.0 C	0.0	10951.5	12178.4	1119.9	0.0
		1.00	1227.0 C	0.0	25168.7	26395.7	1211.5	0.0
	2	.0	57.8 C	0.0	1674.1	1731.9	0.0	0.0
		1.00	57.8 C	0.0	1674.1	1731.9	0.0	0.0
	3	.0	1284.8 C	0.0	12625.6	13910.3	1119.9	0.0
		1.00	1284.8 C	0.0	26842.8	28127.6	1211.5	0.0

***** END OF LATEST ANALYSIS RESULT *****

48. FINISH

***** END OF STAAD-III *****

**** DATE= OCT 22,1997 TIME= 9: 5:15 ****

* For questions on STAAD-III, contact: *

* Research Engineers, Inc at *

* Ph: (714) 974-2500 Fax: (714) 921-2543 *

```

*****
*                               *
*      S T A A D - III         *
*      Revision 20.2           *
*      Proprietary Program of  *
*      Research Engineers, Inc. *
*      Date=   SEP 30, 1997    *
*      Time=   10:42:16        *
*                               *
*      USER ID: SARGENT & LUNDY ENGRS
*****

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1. STAAD PLANE 3X4.19 CHANNEL
2. INPUT WIDTH 72
3. UNIT INCHES POUND
4. JOINT COORDINATES
5. 1 0. 0.; 2 0. .774; 3 0. 1.325; 4 0.551 1.325; 5 2.176 1.325
6. 6 2.727 1.325; 7 2.727 0.774; 8 2.727 0.
7. MEMBER INCIDENCES
8. 1 1 2; 2 2 3; 3 3 4; 4 4 5; 5 5 6; 6 6 7; 7 7 8
9. MEMBER PROPERTY AMERICAN
10. 1 2 PRI IX .1 IY .0016955 IZ .0016955 AY .273 AZ .2 YD .273 ZD 1
11. 3 4 5 PRI IX .1 IY .0004094 IZ .0004094 AY .170 AZ .2 YD .170 ZD 1
12. 6 7 PRI IX .1 IY .0016955 IZ .0016955 AY .273 AZ .2 YD .273 ZD 1
13. CONSTANTS
14. 5 STEEL ALL
15. POISSON STEEL ALL
16. DENSITY STEEL ALL
17. BETA 0. ALL
18. * STAINLESS STEEL
19. ALPHA 99E-7 ALL
20. SUPPORTS
21. 1 8 PINNED
22. LOAD 1 LC 4,5,7
23. JOINT LOAD
24. 3 FX 16.75 FY -21.99
25. 6 FX -10.523 FY -21.99
26. MEMBER LOAD
27. 1 2 UNI GX 197.03
28. 3 4 5 UNI GY -161.13
29. 6 7 UNI GX -123.8
30. LOAD 2 JET IMPINGEMENT
31. JOINT LOAD
32. 3 FY -63.75
33. 6 FY -63.75
34. MEMBER LOAD
35. 3 4 5 UNI GY -467.0
36. LOAD 3 THERMAL
37. TEMP LOAD
38. 3 4 5 TEMP 76.
39. LOAD COMB 4
40. 1 1. 3 1.
41. LOAD COMB 5 JET IMPINGEMENT + THERMAL

3X4.19 CHANNEL

-- PAGE NO. 2

42. 2 1. 3 0.40
43. LOAD LIST ALL
44. PERFORM ANALYSIS

P R O B L E M S T A T I S T I C S

NUMBER OF JOINTS/MEMBER+ELEMENTS/SUPPORTS = 8/ 7/ 2
ORIGINAL/FINAL BAND-WIDTH = 1/ 1
TOTAL PRIMARY LOAD CASES = 3, TOTAL DEGREES OF FREEDOM = 20
SIZE OF STIFFNESS MATRIX = 120 DOUBLE PREC. WORDS
REQD/AVAIL. DISK SPACE = 12.01/ 1070.8 MB, EXMEM = 1.02 MB

++ PROCESSING ELEMENT STIFFNESS MATRIX.	10:42:17
++ PROCESSING GLOBAL STIFFNESS MATRIX.	10:42:17
++ PROCESSING TRIANGULAR FACTORIZATION.	10:42:18
++ CALCULATING JOINT DISPLACEMENTS.	10:42:18
++ CALCULATING MEMBER FORCES.	10:42:18

45. PRINT JOINT DISP LIST 3 6

3x4.19 CHANNEL

-- PAGE NO. 3

JOINT DISPLACEMENT (INCH RADIANS) STRUCTURE TYPE = PLANE

JOINT	LOAD	X-TRANS	Y-TRANS	Z-TRANS	X-ROTAN	Y-ROTAN	Z-ROTAN
3	1	0.00245	-0.00004	0.00000	0.00000	0.00000	-0.00200
	2	0.00006	-0.00012	0.00000	0.00000	0.00000	-0.00251
	3	-0.00102	0.00000	0.00000	0.00000	0.00000	0.00072
	4	0.00143	-0.00004	0.00000	0.00000	0.00000	-0.00129
	5	-0.00035	-0.00012	0.00000	0.00000	0.00000	-0.00223
6	1	0.00235	-0.00004	0.00000	0.00000	0.00000	-0.00081
	2	-0.00006	-0.00012	0.00000	0.00000	0.00000	0.00251
	3	0.00102	0.00000	0.00000	0.00000	0.00000	-0.00072
	4	0.00337	-0.00004	0.00000	0.00000	0.00000	-0.00152
	5	0.00035	-0.00012	0.00000	0.00000	0.00000	0.00223

***** END OF LATEST ANALYSIS RESULT *****

46. PRINT MEMBER FORCES LIST 1 4 7

3X4.19 CHANNEL

-- PAGE NO. 4

MEMBER END FORCES STRUCTURE TYPE = PLANE

ALL UNITS ARE -- POUN INCH

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
1	1	1	215.09	86.47	0.00	0.00	0.00	0.00
		2	-215.09	66.03	0.00	0.00	0.00	7.91
	2	1	700.50	-201.91	0.00	0.00	0.00	0.00
		2	-700.50	201.91	0.00	0.00	0.00	-156.28
	3	1	0.00	-4.70	0.00	0.00	0.00	0.00
		2	0.00	4.70	0.00	0.00	0.00	-3.64
	4	1	215.09	81.76	0.00	0.00	0.00	0.00
		2	-215.09	70.74	0.00	0.00	0.00	4.27
	5	1	700.50	-203.79	0.00	0.00	0.00	0.00
		2	-700.50	203.79	0.00	0.00	0.00	-157.74
4	1	4	191.35	104.32	0.00	0.00	0.00	-23.55
		5	-191.35	157.52	0.00	0.00	0.00	-19.67
	2	4	201.91	379.44	0.00	0.00	0.00	-12.43
		5	-201.91	379.44	0.00	0.00	0.00	12.43
	3	4	4.70	0.00	0.00	0.00	0.00	6.23
		5	-4.70	0.00	0.00	0.00	0.00	-6.23
	4	4	196.05	104.32	0.00	0.00	0.00	-17.32
		5	-196.05	157.52	0.00	0.00	0.00	-25.90
	5	4	203.79	379.44	0.00	0.00	0.00	-9.93
		5	-203.79	379.44	0.00	0.00	0.00	9.93
7	1	7	268.29	112.61	0.00	0.00	0.00	50.08
		8	-268.29	-16.79	0.00	0.00	0.00	0.00
	2	7	700.50	201.91	0.00	0.00	0.00	156.28
		8	-700.50	-201.91	0.00	0.00	0.00	0.00
	3	7	0.00	4.70	0.00	0.00	0.00	3.64
		8	0.00	-4.70	0.00	0.00	0.00	0.00
	4	7	268.29	117.31	0.00	0.00	0.00	53.72
		8	-268.29	-21.49	0.00	0.00	0.00	0.00
	5	7	700.50	203.79	0.00	0.00	0.00	157.74
		8	-700.50	-203.79	0.00	0.00	0.00	0.00

***** END OF LATEST ANALYSIS RESULT *****

47. SECTION 0.5 1. MEMB 4
 48. PRINT SUPPORT REACTIONS

3X4.19 CHANNEL

-- PAGE NO. 5

SUPPORT REACTIONS -UNIT POUN INCH STRUCTURE TYPE = PLANE

JOINT	LOAD	FORCE-X	FORCE-Y	FORCE-Z	MOM-X	MOM-Y	MOM Z
1	1	-86.47	215.09	0.00	0.00	0.00	0.00
	2	201.91	700.50	0.00	0.00	0.00	0.00
	3	4.70	0.00	0.00	0.00	0.00	0.00
	4	-81.76	215.09	0.00	0.00	0.00	0.00
	5	203.70	700.50	0.00	0.00	0.00	0.00
8	1	-16.79	268.29	0.00	0.00	0.00	0.00
	2	-201.91	700.50	0.00	0.00	0.00	0.00
	3	-4.70	0.00	0.00	0.00	0.00	0.00
	4	-21.49	268.29	0.00	0.00	0.00	0.00
	5	-203.79	700.50	0.00	0.00	0.00	0.00

***** END OF LATEST ANALYSIS RESULT *****

49. * UNIT INCH KIP

50. PRINT MEMBER STRESSES LIST 1 4 7

3X4.19 CHANNEL

-- PAGE NO. 6

MEMBER STRESSES

ALL UNITS ARE POUN/SQ INCH

MEMB	LD	SECT	AXIAL	BEND-Y	BEND-Z	COMBINED	SHEAR-Y	SHEAR-Z
1	1	.0	787.9 C	0.0	0.0	787.9	316.7	0.0
		1.00	787.9 C	0.0	636.7	1424.5	241.9	0.0
	2	.0	2565.9 C	0.0	0.0	2566.0	739.6	0.0
		1.00	2565.9 C	0.0	12581.7	15147.7	739.6	0.0
	3	.0	0.0 T	0.0	0.0	0.0	17.2	0.0
		1.00	0.0 T	0.0	293.1	293.1	17.2	0.0
	4	.0	787.9 C	0.0	0.0	787.9	299.5	0.0
		1.00	787.9 C	0.0	343.6	1131.4	259.1	0.0
	5	.0	2565.9 C	0.0	0.0	2566.0	746.5	0.0
		1.00	2565.9 C	0.0	12699.0	15264.9	746.5	0.0
4	1	.0	1125.6 C	0.0	4890.4	6015.9	613.6	0.0
		0.50	1125.6 C	0.0	11445.9	12571.5	156.5	0.0
		1.00	1125.6 C	0.0	4083.4	5209.0	926.6	0.0
	2	.0	1187.7 C	0.0	2579.8	3767.5	2232.0	0.0
		0.50	1187.7 C	0.0	34583.8	35771.6	0.0	0.0
		1.00	1187.7 C	0.0	2579.8	3767.5	2232.0	0.0
	3	.0	27.7 C	0.0	1294.0	1321.6	0.0	0.0
		0.50	27.7 C	0.0	1294.0	1321.6	0.0	0.0
		1.00	27.7 C	0.0	1294.0	1321.6	0.0	0.0
	4	.0	1153.2 C	0.0	3596.4	4749.6	613.6	0.0
		0.50	1153.2 C	0.0	10151.9	11305.2	156.5	0.0
		1.00	1153.2 C	0.0	377.4	6530.6	926.6	0.0
	5	.0	1198.8 C	0.0	2062.2	3261.0	2232.0	0.0
		0.50	1198.8 C	0.0	34066.3	35265.1	0.0	0.0
		1.00	1198.8 C	0.0	2062.2	3261.0	2232.0	0.0
7	1	.0	982.7 C	0.0	4031.6	5014.3	412.5	0.0
		1.00	982.7 C	0.0	0.0	982.7	61.5	0.0
	2	.0	2566.0 C	0.0	12581.7	15147.7	739.6	0.0
		1.00	2566.0 C	0.0	0.0	2566.0	739.6	0.0
	3	.0	0.0 C	0.0	293.1	293.1	17.2	0.0
		1.00	0.0 C	0.0	0.0	0.0	17.2	0.0
	4	.0	982.7 C	0.0	4324.7	5307.4	429.7	0.0
		1.00	982.7 C	0.0	0.0	982.7	78.7	0.0
	5	.0	2566.0 C	0.0	12699.0	15264.9	746.5	0.0
		1.00	2566.0 C	0.0	0.0	2566.0	746.5	0.0

***** END OF LATEST ANALYSIS RESULT *****

51. * PARAMETERS

52. * CODE AISC

53. * CHECK CODE 1 TO 12 14 TO 22

54. FINISH

***** END OF STAAD-III *****

FINAL

**** DATE= SEP 30,1997 . TIME= 10:42:19 ****

```
*****
*      For questions on STAAD-III,  contact:      *
*      Research Engineers, Inc at                  *
*      Ph: (714) 974-2500   Fax:  (714) 921-2543   *
*****
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*****
*                               *
*      S T A A D - III         *
*      Revision 20.2           *
*      Proprietary Program of  *
*      Research Engineers, Inc. *
*      Date=   OCT 28, 1997    *
*      Time=   13:37: 2        *
*                               *
*      USER ID: SARGENT & LUNDY ENGRS *
*****

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1. STAAD PLANE 3X4.19 CHANNEL
2. INPUT WIDTH 72
3. UNIT INCHES POUND
4. JOINT COORDINATES
5. 1 0. .5; 2 0. .774; 3 0. 1.325; 4 0.551 1.325; 5 2.176 1.325
6. 6 2.727 1.325; 7 2.727 0.774; 8 2.727 0.
7. MEMBER INCIDENCES
8. 1 1 2; 2 2 3; 3 3 4; 4 4 5; 5 5 6; 6 6 7; 7 7 8
9. MEMBER PROPERTY AMERICAN
10. 1 2 PRI IX .1 IY .0016955 IZ .0016955 AY .273 AZ .2 YD .273 ZD 1
11. 3 4 5 PRI IX .1 IY .0004094 IZ .0004094 AY .170 AZ .2 YD .170 ZD 1
12. 6 7 PRI IX .1 IY .0016955 IZ .0016955 AY .273 AZ .2 YD .273 ZD 1
13. CONSTANTS
14. E STEEL ALL
15. POISSON STEEL ALL
16. DENSITY STEEL ALL
17. BETA 0. ALL
18. * STAINLESS STEEL
19. ALPHA 99E-7 ALL
20. SUPPORTS
21. 1 8 PINNED
22. LOAD 1 LC 4,5,7 UPPER CHANNEL
23. JOINT LOAD
24. 3 FX 58.48 FY -6.1425
25. 6 FX -3.825 FY -6.1425
26. MEMBER LOAD
27. 1 2 UNI GX 688.
28. 3 4 5 UNI GY -45.
29. 6 7 UNI GX -45
30. LOAD 2 LC 4,5,7 LOWER CHANNEL
31. JOINT LOAD
32. 3 FX 3.825 FY -6.1425
33. 6 FX -58.48 FY -6.1425
34. MEMBER LOAD
35. 1 2 UNI GX 45.
36. 3 4 5 UNI GY -45.
37. 6 7 UNI GX -688.
38. LOAD 3 LC 4,5,7 CHANNEL AT EL. 703'-0"
39. JOINT LOAD
40. 3 FX 3.825 FY -6.1425
41. 6 FX -39.69 FY -6.1425

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3X4.19 CHANNEL

-- PAGE NO. 2

42. MEMBER LOAD
 43. 1 2 UNI GX 45.
 44. 3 4 5 UNI GY -45.
 45. 6 7 UNI GX -467.
 46. LOAD 4 LC 4,5,7 PRESSURE LOAD = 197.03 PSI
 47. JOINT LOAD
 48. 3 FX 10.523 FY -21.99
 49. 6 FX -16.75 FY -21.99
 50. MEMBER LOAD
 51. 1 2 UNI GX 123.8
 52. 3 4 5 UNI GY -161.13
 53. 6 7 UNI GX -197.03
 54. LOAD 5 THERMAL
 55. TEMP LOAD
 56. 1 TO 7 TEMP 76.
 57. LOAD COMB C
 58. 1 1. 5 1.
 59. LOAD COMB 7
 60. 2 1. 5 1.
 61. LOAD COMB 8 CHANNEL AT EL. 703'-0"
 62. 3 1. 5 1
 63. LOAD COMB 9 PRESSURE LOAD + THERMAL
 64. 4 1. 5 1.0
 65. LOAD LIST ALL
 66. PERFORM ANALYSIS

P R O B L E M S T A T I S T I C S

 NUMBER OF JOINTS/MEMBER+ELEMENTS/SUPPORTS = 8/ 7/ 2
 ORIGINAL/FINAL BAND-WIDTH = 1/ 1
 TOTAL PRIMARY LOAD CASES = 5, TOTAL DEGREES OF FREEDOM = 20
 SIZE OF STIFFNESS MATRIX = 120 DOUBLE PREC. WORDS
 REQD/AVAIL. DISK SPACE = 12.01/ 1069.0 MB, EXMEM = 1.02 MB

++ PROCESSING ELEMENT STIFFNESS MATRIX.	13:37: 4
++ PROCESSING GLOBAL STIFFNESS MATRIX.	13:37: 4
++ PROCESSING TRIANGULAR FACTORIZATION.	13:37: 4
++ CALCULATING JOINT DISPLACEMENTS.	13:37: 4
++ CALCULATING MEMBER FORCES.	13:37: 4

67. PRINT JOINT DISP LIST 3 6

3X4.19 CHANNEL

-- PAGE NO. 3

JOINT DISPLACEMENT (INCH RADIANS) STRUCTURE TYPE = PLANE

JOINT	LOAD	X-TRANS	Y-TRANS	Z-TRANS	X-ROTAN	Y-ROTAN	Z-ROTAN
3	1	0.00769	0.00001	0.00000	0.00000	0.00000	-0.00811
	2	-0.01168	-0.00003	0.00000	0.00000	0.00000	0.01242
	3	-0.00765	-0.00002	0.00000	0.00000	0.00000	0.00809
	4	-0.00095	-0.00003	0.00000	0.00000	0.00000	0.00050
	5	-0.00083	0.00062	0.00000	0.00000	0.00000	0.00097
	6	0.00687	0.00063	0.00000	0.00000	0.00000	-0.00714
	7	-0.01250	0.00060	0.00000	0.00000	0.00000	0.01339
	8	-0.00848	0.00060	0.00000	0.00000	0.00000	0.00905
	9	-0.00177	0.00059	0.00000	0.00000	0.00000	0.00147
6	1	0.00760	-0.00003	0.00000	0.00000	0.00000	-0.00425
	2	-0.01189	0.00002	0.00000	0.00000	0.00000	0.00594
	3	-0.00780	0.00001	0.00000	0.00000	0.00000	0.00395
	4	-0.00106	-0.00004	0.00000	0.00000	0.00000	0.00111
	5	0.00122	0.00100	0.00000	0.00000	0.00000	-0.00083
	6	0.00882	0.00097	0.00000	0.00000	0.00000	-0.00508
	7	-0.01067	0.00101	0.00000	0.00000	0.00000	0.00511
	8	-0.00658	0.00100	0.00000	0.00000	0.00000	0.00312
	9	0.00016	0.00096	0.00000	0.00000	0.00000	0.00028

***** END OF LATEST ANALYSIS RESULT *****

68. PRINT MEMBER FORCES LIST 1 4 7

3X4.19 CHANNEL

-- PAGE NO. 4

MEMBER END FORCES STRUCTURE TYPE = PLANE

ALL UNITS ARE -- POUN INCH

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
1	1	1	-49.48	463.69	0.00	0.00	0.00	0.00
		2	49.48	-275.18	0.00	0.00	0.00	101.22
	2	1	241.35	-336.75	0.00	0.00	0.00	0.00
		2	-241.35	349.08	0.00	0.00	0.00	-93.96
	3	1	181.49	-225.46	0.00	0.00	0.00	0.00
		2	-181.49	237.79	0.00	0.00	0.00	-63.46
	4	1	257.07	-92.13	0.00	0.00	0.00	0.00
		2	-257.07	126.06	0.00	0.00	0.00	-29.89
	5	1	-1.34	-7.29	0.00	0.00	0.00	0.00
		2	1.34	7.29	0.00	0.00	0.00	-2.00
	6	1	-50.82	456.40	0.00	0.00	0.00	0.00
		2	50.82	-267.89	0.00	0.00	0.00	99.23
	7	1	240.02	-344.03	0.00	0.00	0.00	0.00
		2	-240.02	356.36	0.00	0.00	0.00	-95.95
	8	1	180.16	-232.74	0.00	0.00	0.00	0.00
		2	-180.16	245.07	0.00	0.00	0.00	-65.46
	9	1	255.74	-99.42	0.00	0.00	0.00	0.00
		2	-255.74	133.34	0.00	0.00	0.00	-31.89
4	1	4	162.39	-80.42	0.00	0.00	0.00	-110.93
		5	-162.39	153.54	0.00	0.00	0.00	-79.17
	2	4	377.70	210.42	0.00	0.00	0.00	170.36
		5	-377.70	-137.29	0.00	0.00	0.00	112.15
	3	4	266.41	150.55	0.00	0.00	0.00	111.53
		5	-266.41	-77.43	0.00	0.00	0.00	73.71
	4	4	204.79	146.30	0.00	0.00	0.00	13.07
		5	-204.79	115.54	0.00	0.00	0.00	11.92
	5	4	7.29	-1.34	0.00	0.00	0.00	6.75
		5	-7.29	1.34	0.00	0.00	0.00	-8.92
	6	4	169.68	-81.75	0.00	0.00	0.00	-104.18
		5	-169.68	154.88	0.00	0.00	0.00	-88.08
	7	4	384.99	209.08	0.00	0.00	0.00	177.11
		5	-384.99	-135.96	0.00	0.00	0.00	103.24
	8	4	273.69	149.22	0.00	0.00	0.00	118.28
		5	-273.69	-76.09	0.00	0.00	0.00	64.79
	9	4	212.08	144.96	0.00	0.00	0.00	19.82
		5	-212.08	116.87	0.00	0.00	0.00	3.00
7	1	7	184.48	133.77	0.00	0.00	0.00	90.06
		8	-184.48	-98.94	0.00	0.00	0.00	0.00
	2	7	-106.35	-59.87	0.00	0.00	0.00	-252.42
		8	106.35	592.38	0.00	0.00	0.00	0.00
	3	7	-46.49	-30.60	0.00	0.00	0.00	-163.57
		8	46.49	392.06	0.00	0.00	0.00	0.00

3X4.19 CHANNEL

-- PAGE NO. 5

MEMBER END FORCES STRUCTURE TYPE = PLANE

ALL UNITS ARE -- POUN INCH

MEMBER	LOAD	JT	AXIAL	SHEAR-Y	SHEAR-Z	TORSION	MOM-Y	MOM-Z
4		7	226.31	79.48	0.00	0.00	0.00	2.50
		8	-226.31	73.02	0.00	0.00	0.00	0.00
5		7	1.34	7.29	0.00	0.00	0.00	5.64
		8	-1.34	-7.29	0.00	0.00	0.00	0.00
6		7	185.82	141.06	0.00	0.00	0.00	95.70
		8	-185.82	-106.23	0.00	0.00	0.00	0.00
7		7	-105.02	-52.58	0.00	0.00	0.00	-246.78
		8	105.02	585.10	0.00	0.00	0.00	0.00
8		7	-45.16	-23.32	0.00	0.00	0.00	-157.93
		8	45.16	384.77	0.00	0.00	0.00	0.00
9		7	227.65	86.76	0.00	0.00	0.00	8.14
		8	-227.65	65.74	0.00	0.00	0.00	0.00

***** END OF LATEST ANALYSIS RESULT *****

69. SECTION 0. .5 1. MEMB 4

70. PRINT SUPPORT REACTIONS

3X4.19 CHANNEL

-- PAGE NO. 6

SUPPORT REACTIONS -UNIT POUN INCH STRUCTURE TYPE = PLANE

JOINT	LOAD	FORCE-X	FORCE-Y	FORCE-Z	MOM-X	MOM-Y	MOM Z
1	1	-463.69	-49.48	0.00	0.00	0.00	0.00
	2	336.75	241.35	0.00	0.00	0.00	0.00
	3	225.46	181.49	0.00	0.00	0.00	0.00
	4	92.13	257.07	0.00	0.00	0.00	0.00
	5	7.29	-1.34	0.00	0.00	0.00	0.00
	6	-456.40	-50.82	0.00	0.00	0.00	0.00
	7	344.03	240.02	0.00	0.00	0.00	0.00
	8	232.74	180.16	0.00	0.00	0.00	0.00
	9	99.42	255.74	0.00	0.00	0.00	0.00
8	1	-98.94	184.48	0.00	0.00	0.00	0.00
	2	592.38	-106.35	0.00	0.00	0.00	0.00
	3	392.06	-46.49	0.00	0.00	0.00	0.00
	4	73.02	226.31	0.00	0.00	0.00	0.00
	5	-7.29	1.34	0.00	0.00	0.00	0.00
	6	-106.23	185.82	0.00	0.00	0.00	0.00
	7	585.10	-105.02	0.00	0.00	0.00	0.00
	8	384.77	-45.16	0.00	0.00	0.00	0.00
	9	65.74	227.65	0.00	0.00	0.00	0.00

***** END OF LATEST ANALYSIS RESULT *****

71. PRINT MEMBER STRESSES LIST 1 4 7

3X4.19 CHANNEL

-- PAGE NO. 7

MEMBER STRESSES

ALL UNITS ARE POUN/SQ INCH

MEMB	LD	SECT	AXIAL	BEND-Y	BEND-Z	COMBINED	SHEAR-Y	SHEAR-Z
1	1	.0	181.2 T	0.0	0.0	181.3	1698.5	0.0
		1.00	181.2 T	0.0	8149.3	8330.5	1008.0	0.0
	2	.0	884.1 C	0.0	0.0	884.1	1233.5	0.0
		1.00	884.1 C	0.0	7564.3	8448.4	1278.7	0.0
	3	.0	664.8 C	0.0	0.0	664.8	825.8	0.0
		1.00	664.8 C	0.0	5109.3	5774.1	871.0	0.0
	4	.0	941.7 C	0.0	0.0	941.7	337.5	0.0
		1.00	941.7 C	0.0	2406.5	3348.2	461.7	0.0
	5	.0	4.9 T	0.0	0.0	4.9	26.7	0.0
		1.00	4.9 T	0.0	160.7	165.6	26.7	0.0
	6	.0	186.1 T	0.0	0.0	186.2	1671.8	0.0
		1.00	186.1 T	0.0	7988.6	8174.7	981.3	0.0
	7	.0	879.2 C	0.0	0.0	879.2	1260.2	0.0
		1.00	879.2 C	0.0	7725.0	8604.2	1305.4	0.0
	8	.0	659.9 C	0.0	0.0	659.9	852.5	0.0
		1.00	659.9 C	0.0	5270.0	5929.9	897.7	0.0
	9	.0	936.8 C	0.0	0.0	936.8	364.2	0.0
		1.00	936.8 C	0.0	2567.2	3504.0	488.4	0.0
4	1	.0	955.3 C	0.0	23030.8	23986.0	473.1	0.0
		0.50	955.3 C	0.0	6380.9	7336.1	688.1	0.0
		1.00	955.3 C	0.0	16436.8	17392.1	903.2	0.0
	2	.0	2221.8 C	0.0	35370.3	37592.1	1237.7	0.0
		0.50	2221.8 C	0.0	2958.5	5180.3	1022.7	0.0
		1.00	2221.8 C	0.0	22285.5	25507.3	807.6	0.0
	3	.0	1567.1 C	0.0	23155.8	24722.8	885.6	0.0
		0.50	1567.1 C	0.0	842.4	2409.5	670.5	0.0
		1.00	1567.1 C	0.0	15303.2	16870.2	455.5	0.0
	4	.0	1204.7 C	0.0	2713.9	3918.6	860.6	0.0
		0.50	1204.7 C	0.0	10923.0	12127.6	90.5	0.0
		1.00	1204.7 C	0.0	2475.0	3679.7	679.6	0.0
	5	.0	42.9 C	0.0	1400.6	1443.5	7.9	0.0
		0.50	42.9 C	0.0	1626.0	1668.8	7.9	0.0
		1.00	42.9 C	0.0	1851.3	1894.2	7.9	0.0
	6	.0	998.1 C	0.0	21630.1	22628.3	480.9	0.0
		0.50	998.1 C	0.0	4754.9	5753.0	696.0	0.0
		1.00	998.1 C	0.0	18288.1	19286.2	911.1	0.0
	7	.0	2264.6 C	0.0	36771.0	39035.6	1229.9	0.0
		0.50	2264.6 C	0.0	4584.5	6849.1	1014.8	0.0
		1.00	2264.6 C	0.0	21434.2	23698.9	799.7	0.0
	8	.0	1609.9 C	0.0	24556.4	26166.3	877.8	0.0
		0.50	1609.9 C	0.0	2468.4	4078.3	662.7	0.0
		1.00	1609.9 C	0.0	13451.9	15061.8	447.6	0.0
	9	.0	1247.5 C	0.0	4114.6	5362.1	852.7	0.0
		0.50	1247.5 C	0.0	9297.0	10544.5	82.6	0.0
		1.00	1247.5 C	0.0	623.7	1871.2	687.5	0.0
7	1	.0	675.8 C	0.0	7250.6	7926.3	490.0	0.0
		1.00	675.8 C	0.0	0.0	675.8	362.4	0.0

3X4.19 CHANNEL

-- PAGE NO. 8/FINAL

MEMBER STRESSES

ALL UNITS ARE POUN/SQ INCH

MEMB	LD	SECT	AXIAL	BEND-Y	BEND-Z	COMBINED	SHEAR-Y	SHEAR-Z
2	.0		389.6 T	0.0	20321.7	20711.3	219.3	0.0
	1.00		389.6 T	0.0	0.0	389.6	2169.9	0.0
3	.0		170.3 T	0.0	13168.5	13338.8	112.1	0.0
	1.00		170.3 T	0.0	0.0	170.3	1436.1	0.0
4	.0		829.0 C	0.0	201.1	1030.1	291.1	0.0
	1.00		829.0 C	0.0	0.0	829.0	267.5	0.0
5	.0		4.9 C	0.0	454.0	458.8	26.7	0.0
	1.00		4.9 C	0.0	0.0	4.9	26.7	0.0
6	.0		680.6 C	0.0	7704.5	8385.2	516.7	0.0
	1.00		680.6 C	0.0	0.0	680.7	389.1	0.0
7	.0		384.7 T	0.0	19867.8	20252.5	192.5	0.0
	1.00		384.7 T	0.0	0.0	384.7	2143.2	0.0
8	.0		165.4 T	0.0	12714.6	12880.0	85.4	0.0
	1.00		165.4 T	0.0	0.0	165.4	1409.4	0.0
9	.0		833.9 C	0.0	655.1	1489.0	317.8	0.0
	1.00		833.9 C	0.0	0.0	833.9	240.8	0.0

***** END OF LATEST ANALYSIS RESULT *****

72. FINISH

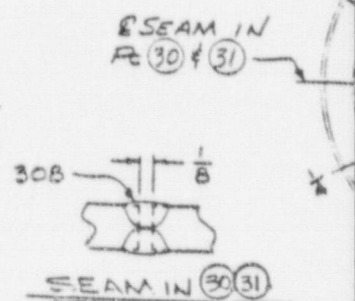
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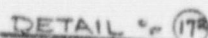
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* For questions on STAAD-III, contact: *

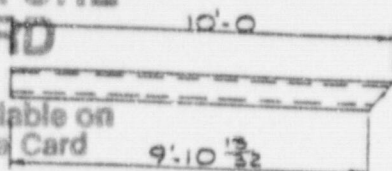
* Research Engineers, Inc at *

* Ph: (714) 974-2500 Fax: (714) 921-2543 *

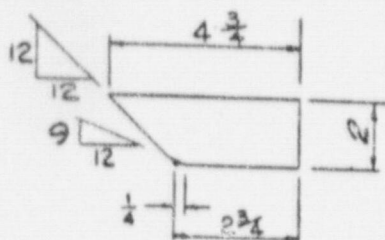




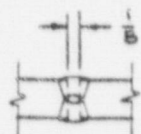
Also Available on
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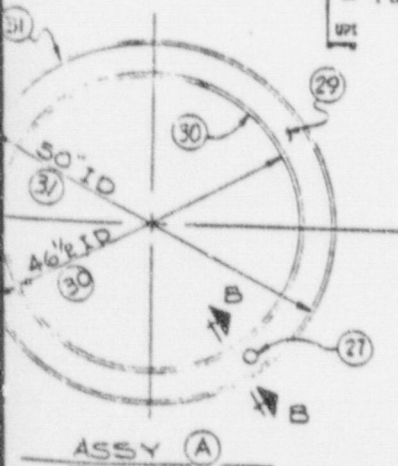


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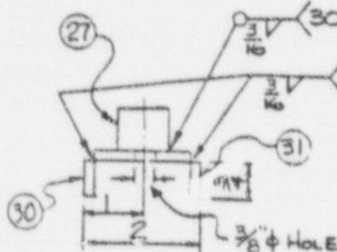


TYR SHOP SPLICE
FOR CHANNELS

R/c



ASSY (A)



SECTION B-B

▶ INDICATES CHANGE FROM PREVIOUS ISSUE

LINE #	MARK	ASME #	DESCRIPTION	LENGTH		SPEC	#	AB	LN
				FT	IN				
2	23-1		CHANNEL 2½ x 1 x ¼ (SK)	17	5 ¾	A36	D2	8/5	
4	23-2		CHANNEL 2½ x 1 x ¼ (ROLL HARD WAY)	FIN SHR	15 9 ½ 9	do.	D2	8/6	
4	23-3		CHANNEL 2 x 1 x ¾	1	4 ¾	do.	D2	8/20	
4	23-4		CHANNEL 2 x 1 x ¾	1	4 ¾	do.	D2	8/20	
8	23-5		LSK x ¼ (9" L 2½ x 2½ x ¼ x 0'-4 ½)			A36	D2	7/6	
4	23-37		L 5 x 2 x ½ ¼ (L 5 x 3 x ½)	FIN	18 10 ½	A36	D2		
			FIN RAD = 12' 0 ¼	SHR	20 10 ½				
6	23-36		CHANNEL C 3 x 5.0	1	3	A36	D2	9/6	
4	23-39		L 5 x 2 x ½ ¼ (L 5 x 3 x ½)	FIN	20 10 ½	A36	D2		
			FIN RAD = 13' 3 ½	SHR	22 10 ½				
4	23-9		CHANNEL 2 x 1 x ¾	1	2 ¾	do.	D2	8/20	
4	23-10		CHANNEL 2 x 1 x ¾	1	2 ¾	do.	D2	8/20	
6	23-11		CHANNEL 2½ x 1 x ¼	FIN	15 11 ½	do.	D2	8/4	
			(ROLL HARD WAY)	SHR	17 11				
4	23-12 ^R		CHANNEL 2 x 1 x ¾	SK OF	22 10 ½	do.	D2	8/6	
4	23-13 ^R		CHANNEL 2 x 1 x ¾	SK OF	20 11 ½	do.	D2	8/6	
4	23-14		CHANNEL 2 x 1 x ¾		13 9 ½	do.	D2	8/10	
4	23-15 ^R		CHANNEL 2 x 1 x ¾	SK OF	16 2 ½	do.	D2	8/11	
4	23-16 ^R		CHANNEL 2½ x 1 x ¼	SK OF	8 0 ½	do.	D2	8/12	
4	23-17 ^R		CHANNEL 2 x 1 x ¾	SK OF	23 7 ½	do.	D2	8/13	
8	23-18		CHANNEL 2 x 1 x ¾		9 1	do.	D2	8/14	
4	23-19 ^R		CHANNEL 2 x 1 x ¾	SK OF	10 0	do.	D2	8/15	
4	23-20 ^R		CHANNEL 2 x 1 x ¾	SK OF	18 7 ½	do.	D2	8/16	
2	23-21		CHANNEL 2 x 1 x ¾		20 5	do.	D2	8/17	
2	23-22		CHANNEL 2 x 1 x ¾		18 4	do.	D2	8/18	
12	23-23		L 2½ x 1½ x ¼	FIN	21 10 ¼	do.	D2	8/19	
			FIN RAD = 41'-9	SHR	23 10 ¼				
3	23-24		R 5K x ¼ (9" R 2 x 0'-4 ¾ 9")			do.	D2	8/20	
4	23-25		R 2 x ¼		0 3	do.	D2	8/21	
4	23-26		R 3½ x ¼		0 5½	A36	D2		
15	23-27		½" φ 3000" HALF CPLG (SCR'D)			do.	D2	8/22	
15	23-28		½" φ 3000" SQUARE HEAD PLUG			do.	D2	8/23	
18	23-A		COLUMN BASE LEAK TEST SYSTEM						
	23-29	18	R 50" O.D. x ¼ x 47" I.D.			do.	D2	8/24	
	23-30	18	R ¾ x ¼ ROLL TO CYLINDER (ROLL RAD = 1'-1¼)	12	2¾	do.	D2	8/25	
	23-31	18	R ¾ x ¼ ROLL TO CYLINDER (ROLL RAD = 2'-1)	13	1¾	do.	D2	8/26	
	23-27	18	½" φ 3000" HALF CPLG (SCR'D)			do.	D2	8/27	
	23-28	18	½" φ 3000" SQUARE HEAD PLUG			do.	D2	8/28	
18	23-B		COLUMN BASE LEAK TEST SYSTEM						
	23-32	18	R 20½" O.D. x ¼ x 17½" I.D.			A36	D2	8/29	
	23-33	18	R ¾ x ¼ ROLL TO CYLINDER (ROLL RAD = 1'-1¼)	4	6½	do.	D2	8/30	
	23-34	18	R ¾ x ¼ ROLL TO CYLINDER (ROLL RAD = 10'3)	5	5½	do.	D2	8/32	
	23-35	18	½" φ 3000" HALF CPLG (SCR'D)			do.	D2		
	23-36	18	½" φ 3000" SQUARE HEAD PLUG			do.	D2		

WORK THIS DWG WITH DWG 2 & 22
AFK 29 1973

SARGENT

97120200

Chicago Bridge & Iron Company

SHOP DETAILS FOR LEAK
TEST SYSTEM IN BOTTOM OF
SUPPRESSION CHAMBER

Purchase's No. 160496

a. Th. ^{red} ~~subsp. nov.~~ 11-19-72

Carroll et al.

73-6336

Page 12 of 12

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MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

Piece-Mark	Serial No.	Material Heat No.	Matl. Type	Piece-Mark	Serial No.	Material Heat No.	Matl. Type
23-3	4PCS	40161					
23-4	4PCS	40161					
23-5	8PCS	14592					
23-6	4PCS	74800					
23-8	4PCS	74800					
23-9	4PCS	40161					
23-10	4PCS	40161					
23-22	2PCS	23755					
23-21	2PCS	23755					
23-15	4PCS	23755					
23-18	2PCS	F30188					
23-18	6PCS	42379					
23-13	4PCS	40161					
23-19	4PCS	F30188					
23-17	4PCS	40161					
23-12	4PCS	40161					
23-23	12PCS	68940					
23-2	4PCS	30844					
23-11	6PCS	30844					
23-1	2PCS	30844					
23-14	4PCS	30706					
23-20	4PCS	30706					
23-16	4PCS	30706					

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date 10/8/74

REV	BY
	CHKD
	DATE

Reviewed (for material covered by code):

Authorized Inspector (Shop)

Date _____

9-1

Contract No.
73-6336

No. 23
Sh 4 of

CB 5 I

CMT R's

Paper Count 53



REYNOLDS ALUMINUM SUPPLY COMPANY
METALS AND BUILDING PRODUCTS

4500 FIFTH AVENUE, S. - P. O. BOX 10785 - BIRMINGHAM, ALABAMA 35202 - 205/591-2341

September 31, 1974

Chicago Bridge
P. O. Box 774
Kankakee, Illinois 60901

Attention: Art Johnson

Please use this letter as authorization to mark material on P.O.
C73-6336-12A as follows:

6 pcs.	Ht. # F30188
22 pcs.	Ht. # 23755
39 pcs.	Ht. # 17797
7 pcs.	Ht. # 42379

Very truly yours,

REYNOLDS ALUMINUM SUPPLY CO.

Bob Littleton

BL:rf

cc: File

CERTIFICATE OF TEST
STAINLESS STEEL

Customer Chicago Bridge Date 8-19-74
Address _____ Customer Order No. 73-6336-12A
Consignee _____ Consignee Order No. _____
Address _____
Invoice No. 07-07219 Date of Shipment 8-2-74
Specification ASTM-A-240 Grade 304

MATERIAL

ITEM	HEAT NUMBER	NUMBER OF PIECES	SIZE	CONDITION
1.	F30188	6	2 X 1 X 3/16 X 120	23-18, 23-
2.	23755	11	2 X 1 X 3/16 X 120	23-21, 23-15, 23-2
3.	23755	11	2 X 1 X 3/16 X 120	
4.	17797	39	2 X 1 X 3/16 X 120	23-18
5.	42379	7	2 X 1 X 3/16 X 120	

CHEMICAL ANALYSES

ITEM	C	MN	P	S	SI	CR	NI	CU	TI	CB	MO		
1.	.046	1.62	.021	.011	.56	18.18	8.79						
2.	.04	1.65	.026	.016	.70	18.30	8.60						
3.	.04	1.65	.026	.016	.70	18.30	8.60						
4.	.033	1.62	.025	.014	.25	18.15	9.10						
5.	.05	1.47	.019	.011	.51	18.22	8.60						

PHYSICAL PROPERTIES

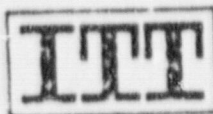
ITEM	TENSILE STRENGTH	YIELD STRENGTH	% ELONGATION	HARDNESS	REMARKS
1.	87,200	39,000	50	686	
2.	80,800	39,700	59.1	RB 82	
3.	80,800	39,700	59.1	RB 82	
4.	85,200	38,200	62	B78-79	
5.	86,700	38,100	56.5	RB 82	

We certify that the chemical analyses and physical test results applying on the above order number are correct and true to the best of our knowledge and belief

Sworn and subscribed to before me

Reynolds Aluminum Supply Company

Diane Smith



ITT Harper Inc.
8200 Lehigh Avenue, Morton Grove Ill 60053
(312) 966-8000

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 6 of 39 R/O

CERTIFIED TEST REPORT

Customer:

METAL GOODS SERVICE CENTER
DIV ALCAN ALUM
11400 W. ADDISON STREET
FRANKLIN PARK, ILL. 60131

Address:

Date: 6/14/74

HMH Register No. 005439

HMH Drawing No.

HMH Work Order No.

Purchase Order No. 24-91055

Description of Item				
Weight and Pieces	Shape	Size	Length	Material
1090 LBS 13 PCS	CHANNEL	2 1/2 x 1 x 1/4"	18/24'	304 STAINLESS STEEL

* SPEC ASTM A276

23-11

CBI C-73-6331
QIR NUMBER 21-91055
DATE SHIPPED 6-20-74
SPECIFICATIONS ASTM A276
ITEM T304 SS Chan

Chemical Analysis									
Heat Number	C	S	P	CU	MO	CR	NI	SI	MN
30844	.064	.028	.021	.21	.29	18.65	9.91	.22	1.38

2 1/2 x 1 x 1/4 18'

SPRAN 7/23/

Mechanical Properties					
Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
30844	88,300	50,000	49%	69	

We hereby certify that the above figures are correct as contained in the records of this company.

Louis Paul, O.C. MGR.



ITT Harpor Inc. 
8200 Lehigh Avenue, Morton Grove, Ill. 60053
(312) 668-6000

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 7 of 39 R/c

CERTIFIED TEST REPORT

Date: 6/4/74

CHANGED
ITCH

HMH Register No. 005090

HMH Drawing No.

HMH Work Order No.

Customer: METAL GOODS SERVICE
DIV ALCAN ALUM
P.O. BOX 427
FRAZLIN PARK, ILL. 60131

Order No. 24-90807

Description of Item

Weight and Pieces	Shape	Size	Length	Material
8045 LBS 110 PCS	CHANNEL	2 x 1 x 1/4"	R/L	Chicago Bridge & Iron 304 STAINLESS STEEL
				C 73-6337-55
				OUR M 1155 24-90985

DATE SHIPPED 6-7-74
SPECIFICATIONS ASTM A 236
ITEM I 304 S/S Channel

23-16
23-20

Chemical Analysis

Heat Number	C	S	P	CU	MO	CR	NI	SI	MN
30599	.000	.012	.022	.43	.25	18.38	9.65	.45	1.45
30706	.064	.021	.023	.31	.34	18.69	10.03	.24	1.43
30013	.049	.018	.023	.26	.28	18.68	9.50	.34	1.45

2 x 1 x 1/4
Cindy Zwickler 6-26-74
10/24/74

Mechanical Properties

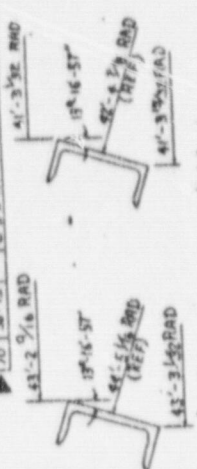
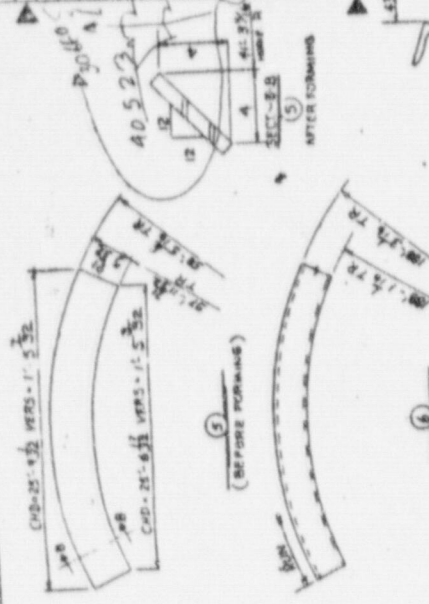
Heat Number	Tensile Strength	Yield Strength	% Elongation	R of A	Hardness	PCS	LDS
30599	90,600	51,700	50	68		57	305
30706	91,000	50,800	48	67		59	402
30013	89,200	49,300	51	70		2	13

We hereby certify that the above figures are correct as contained in the records of this company.

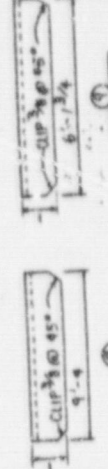
Louis Fred. O.C. MGR.

13
10
0

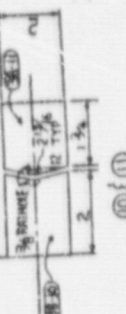
STATION	DESCRIPTION	AMOUNT	DATE
138-1	2 X 12 X 16	10.58	5/1/75
138-2	2 X 12 X 16	10.58	5/1/75
138-3	2 X 12 X 16	10.58	5/1/75
138-4	2 X 12 X 16	10.58	5/1/75
138-5	2 X 12 X 16	10.58	5/1/75
138-6	2 X 12 X 16	10.58	5/1/75
138-7	2 X 12 X 16	10.58	5/1/75
138-8	2 X 12 X 16	10.58	5/1/75
138-9	2 X 12 X 16	10.58	5/1/75
138-10	2 X 12 X 16	10.58	5/1/75
138-11	2 X 12 X 16	10.58	5/1/75
138-12	2 X 12 X 16	10.58	5/1/75
138-13	2 X 12 X 16	10.58	5/1/75
138-14	2 X 12 X 16	10.58	5/1/75
138-15	2 X 12 X 16	10.58	5/1/75
138-16	2 X 12 X 16	10.58	5/1/75
138-17	2 X 12 X 16	10.58	5/1/75
138-18	2 X 12 X 16	10.58	5/1/75
138-19	2 X 12 X 16	10.58	5/1/75
138-20	2 X 12 X 16	10.58	5/1/75
138-21	2 X 12 X 16	10.58	5/1/75
138-22	2 X 12 X 16	10.58	5/1/75
138-23	2 X 12 X 16	10.58	5/1/75
138-24	2 X 12 X 16	10.58	5/1/75
138-25	2 X 12 X 16	10.58	5/1/75
138-26	2 X 12 X 16	10.58	5/1/75
138-27	2 X 12 X 16	10.58	5/1/75
138-28	2 X 12 X 16	10.58	5/1/75
138-29	2 X 12 X 16	10.58	5/1/75
138-30	2 X 12 X 16	10.58	5/1/75
138-31	2 X 12 X 16	10.58	5/1/75
138-32	2 X 12 X 16	10.58	5/1/75
138-33	2 X 12 X 16	10.58	5/1/75
138-34	2 X 12 X 16	10.58	5/1/75
138-35	2 X 12 X 16	10.58	5/1/75
138-36	2 X 12 X 16	10.58	5/1/75
138-37	2 X 12 X 16	10.58	5/1/75
138-38	2 X 12 X 16	10.58	5/1/75
138-39	2 X 12 X 16	10.58	5/1/75
138-40	2 X 12 X 16	10.58	5/1/75
138-41	2 X 12 X 16	10.58	5/1/75
138-42	2 X 12 X 16	10.58	5/1/75
138-43	2 X 12 X 16	10.58	5/1/75
138-44	2 X 12 X 16	10.58	5/1/75
138-45	2 X 12 X 16	10.58	5/1/75
138-46	2 X 12 X 16	10.58	5/1/75
138-47	2 X 12 X 16	10.58	5/1/75
138-48	2 X 12 X 16	10.58	5/1/75
138-49	2 X 12 X 16	10.58	5/1/75
138-50	2 X 12 X 16	10.58	5/1/75
138-51	2 X 12 X 16	10.58	5/1/75
138-52	2 X 12 X 16	10.58	5/1/75
138-53	2 X 12 X 16	10.58	5/1/75
138-54	2 X 12 X 16	10.58	5/1/75
138-55	2 X 12 X 16	10.58	5/1/75
138-56	2 X 12 X 16	10.58	5/1/75
138-57	2 X 12 X 16	10.58	5/1/75
138-58	2 X 12 X 16	10.58	5/1/75
138-59	2 X 12 X 16	10.58	5/1/75
138-60	2 X 12 X 16	10.58	5/1/75
138-61	2 X 12 X 16	10.58	5/1/75
138-62	2 X 12 X 16	10.58	5/1/75
138-63	2 X 12 X 16	10.58	5/1/75
138-64	2 X 12 X 16	10.58	5/1/75
138-65	2 X 12 X 16	10.58	5/1/75
138-66	2 X 12 X 16	10.58	5/1/75
138-67	2 X 12 X 16	10.58	5/1/75
138-68	2 X 12 X 16	10.58	5/1/75
138-69	2 X 12 X 16	10.58	5/1/75
138-70	2 X 12 X 16	10.58	5/1/75
138-71	2 X 12 X 16	10.58	5/1/75
138-72	2 X 12 X 16	10.58	5/1/75
138-73	2 X 12 X 16	10.58	5/1/75
138-74	2 X 12 X 16	10.58	5/1/75
138-75	2 X 12 X 16	10.58	5/1/75
138-76	2 X 12 X 16	10.58	5/1/75
138-77	2 X 12 X 16	10.58	5/1/75
138-78	2 X 12 X 16	10.58	5/1/75
138-79	2 X 12 X 16	10.58	5/1/75
138-80	2 X 12 X 16	10.58	5/1/75
138-81	2 X 12 X 16	10.58	5/1/75
138-82	2 X 12 X 16	10.58	5/1/75
138-83	2 X 12 X 16	10.58	5/1/75
138-84	2 X 12 X 16	10.58	5/1/75
138-85	2 X 12 X 16	10.58	5/1/75
138-86	2 X 12 X 16	10.58	5/1/75
138-87	2 X 12 X 16	10.58	5/1/75
138-88	2 X 12 X 16	10.58	5/1/75
138-89	2 X 12 X 16	10.58	5/1/75
138-90	2 X 12 X 16	10.58	5/1/75
138-91	2 X 12 X 16	10.58	5/1/75
138-92	2 X 12 X 16	10.58	5/1/75
138-93	2 X 12 X 16	10.58	5/1/75
138-94	2 X 12 X 16	10.58	5/1/75
138-95	2 X 12 X 16	10.58	5/1/75
138-96	2 X 12 X 16	10.58	5/1/75
138-97	2 X 12 X 16	10.58	5/1/75
138-98	2 X 12 X 16	10.58	5/1/75
138-99	2 X 12 X 16	10.58	5/1/75
139-00	2 X 12 X 16	10.58	5/1/75



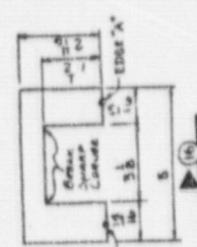
SECTION 7 (AFTER FORMING)
(DIMS. ARE ON THE WHEEL OF CHANNEL)



RECEIVED
MAR 3 1975
SAPGENT
LUNDY

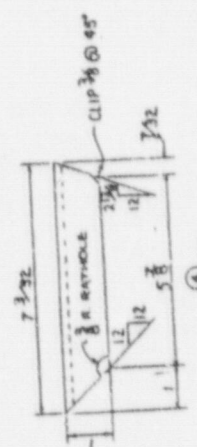
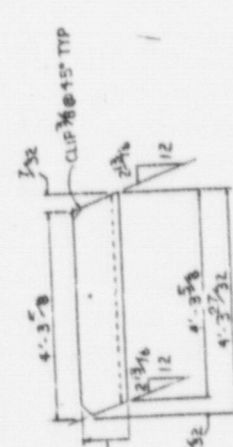
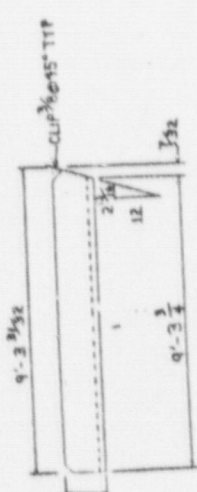


LA SALLE COUNTY STATION, UNIT 1
COMMERCIAL EDITION CONTACT



Chicago Bridge & Iron Company
SUPP CHAMBERS LEAK
TEST DETAILS
Per S/L D-4-5424, April 3
73-6336

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 8 of 39 R/c





MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

Piece-Mark	Serial No.	Material Heat No.	Matl. Type	Piece-Mark	Serial No.	Material Heat No.	Matl. Type
3B-1	IPC	40523				Calculation L-001337	
3B-6	IPC	40523				Project No. 10248-012	
3B-7	IPC	40523				ATTACHMENT D	
						Page 9 of 39 R/O	

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date _____

1/8/75

REV

BY

CHKD

DATE _____

Reviewed (for material covered by code):

Authorized Inspector (Shop)

N/A

Date _____

17-1

Contract No. _____

73-6336

No. 38

Sh 1 of



MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

[illegible]

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 10 of 39 *2/0*

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date _____

11/21/75

7536

BY

CHKD

DATE _____

Reviewed (for material covered by code):

Authorized Inspector (Shop)

N/A

D

17-1

Contract No. _____

73-6336

No. 38

Sh 3 of



MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

[illegible]

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date _____

1/14/75

REV

BY

CHKD

DATE _____

Reviewed (for material covered by code):

Authorized Inspector (Shop)

2/2

Date _____

Contract No. _____

73-6336

† 10.

38

St. 2 of

QUN P.O. NO.

C73-6331-31

QUN NUMBER

24-91053

DATE SHIPPED

11/20/74

SPECIFICATIONS

ASTM A 276

ITEM

T 304 Channel

2 x 1 x 3/16

18/24 R/L

12/30/77

64 PCS

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 12 of 39 R/C

CERTIFIED TEST REPORT

Date: 11/12/74

HMH Register No. 005727

HMH Drawing No. 8484

HMH Work Order No. AA 686

RS

50131

Description of Item

Weight and Pieces	Shape	Size	Length	Material
3610 LBS 64 PCS	CHANNEL	2x1x3/16	20' R/L	304 STAINLESS STEEL

SPEC: ASTM-A-276

Chemical Analysis

Heat Number	C	S	P	CU	NO	CR	NI	SI	MN
40523	.073	.020	.023	.36	.20	18.00	10.02	.31	1.36

Mechanical Properties

Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
40523	23,600	15,100	40.0	75.0	

We hereby certify that the above figures are correct as contained in the records of this company.

BY John P. Pomeroy S.C. Manager
Name and Title

**ITT Harper Inc.**8200 Lehigh Avenue
(312) 966-6000

OUR NUMBER

DATE SHIPPED

3-5-74

SPECIFICATIONS

ASTM-A-236

ITEM

① T304 SS Channel
2 x 1 x 3/16

CERTIFIED TEST REPORT

Date: 3/4/74

HMH Register No.

00130

Braun
4-5-74Customer: METAL GOODS SERVICE CENTER
DIV. ALCAN ALUMINUM
11400 W. ADDISON STREET
Address: FRANKLIN PARK, ILLINOIS 60131Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 13 of 39 R/O

HMH Drawing No.

HMH Work Order No.

Purchase Order No. 74-30675

Description of Item

Weight and Pieces	Shape	Size	Length	Material
3315 LBS 61 PCS	CHANNEL	2 x 1 x 3/16	19' 1/2	304 STAINLESS STEEL
920 LBS 11 PCS	CHANNEL	3 x 1 3/8 x 3/16	1'-24 1/2	304 STAINLESS STEEL

Chemical Analysis

Heat Number	C	S	P	CU	MO	CR	NI	SI	MN	LBS	PCS
30660	.006	.023	.023	.30	.29	13.84	9.7	.57	1.41	2730	51
30549	.52	.015	.025	.23	.27	13.56	9.90	.46	1.60	585	10
40079	.04	.014	.021	.20	.19	12.55	9.63	.53	1.83	22	1

Mechanical Properties

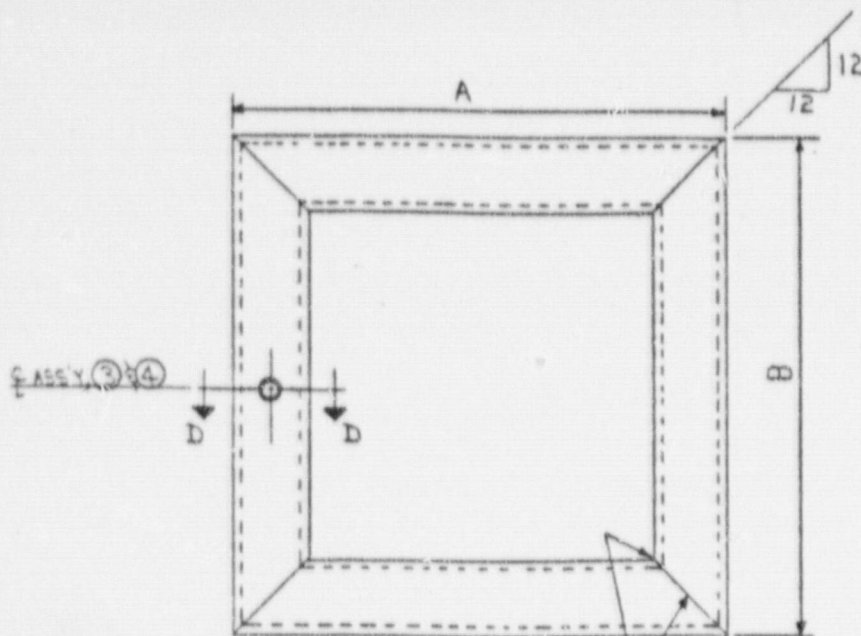
Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
30660	20,850	55,100	42	61.6	
30549	86,400	43,750	46.8	69	
40079	85,350	40,200	52.5	71	

We hereby certify that the above figures are correct as contained in the records of this company.

By

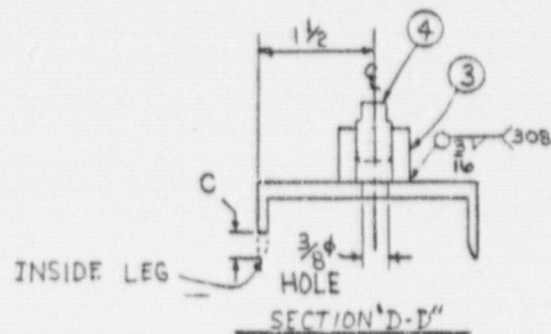
James Taul

Name and Title



DETAIL "E" (TYP)

LEAK TEST ASS'Y
INSIDE ELEV VIEW



DETAIL "E"

SHOP NOTE! THIS WELD MUST
BE LEAK TIGHT @ 4 CORNERS
ON EACH ASS'Y

ASSY	TEST CPLGR	HORIZ R. DIM "A"	VERT P. DIM "B"	C	LEAK TEST FOR ASSY
A	YES	1'-1 1/2	1'-1 1/2	1/2	79-A
B	NO	1'-1 1/2	1'-6 1/4	1/2	81-B
C	NO	1'-4 1/16	2'-10 1/4	1/2	82-A
D	NO	1'-1 1/2	1'-4 1/4	1/2	90-B
E	NO	1'-8 1/2	2'-4 1/2	1/2	89-A
F	NO	1'-11 3/16	2'-4 1/4	1/2	91-B
G	NO	1'-10 7/8	2'-1 1/4	1/2	88-A
H	NO	1'-13 1/16	2'-1 1/2	1/2	85-A
J	NO	1'-11 1/2	2'-1 1/2	1/2	88-B
K	NO	1'-1 1/2	1'-1 1/4	1/2	83B & 89-B
M	YES	1'-10 7/8	1'-1 1/2	1/2	85-B
P	NO	1'-10 3/16	2'-1 1/4	1/2	89-A
R	NO	1'-3	1'-1 1/2	1/2	81-A
S	NO	1'-1 1/2	1'-7 1/4	1/2	87-B
T	NO	1'-1 1/2	1'-9 1/4	1/2	86-B
W	NO	1'-7 1/8	2'-4 1/4	1/2	87-A
Y	NO	1'-10 7/8	1'-1 1/4	1/2	86-A
AA	NO	1'-8 5/8	1'-1 1/2	1/2	82-B
BB	NO	1'-7 7/8	2'-10 1/2	1/2	92-A
CC	NO	1'-6 11/32	2'-4 1/2	1/2	91-A
DD	YES	1'-2 1/2	1'-2 1/2	1/2	75-B
EE	YES	11	11	0	75-A
FF	NO	1'-5 1/2	1'-11 1/2	1/2	76-A


**MATERIAL HEAT
NUMBER SHEET**

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

Piece- Mark	Serial No.	Material Heat No.	Matl. Type	Piece- Mark	Serial No.	Material Heat No.	Matl. Type
39A	32 ASSY'S.	LEAK TEST ASSY					
✓ 39-1	32 PCS.	40489				Calculation L-001337	
✓ 39-2	96 PCS.	40489				Project No. 10248-012	
✓ 39-3	32 PCS.	JR				ATTACHMENT D	
✓ 39-4	32 PCS.	KY				Page 15 Of 39 R/C	
39B	2 ASSY'S	LEAK TEST ASSY					
✓ 39-2	4 PCS	40489					
✓ 39-5	4 PCS	40489					
39C	1 ASSY	LEAK TEST ASSY					
✓ 39-6	2 PCS	40489					
✓ 39-7	2 PCS.	40489					
39D	4 ASSY'S.	LEAK TEST ASSY.					
✓ 39-2	8 PCS	40489					
39-8	8 PCS.	40489					
39E	1 ASSY.	LEAK TEST ASSY.					
✓ 39-9	2 PCS.	40489					
✓ 39-10	2 PCS.	40489					
39F-F	30 ASSY'S.	LEAK TEST ASSY					
✓ 39-31	60 PCS.	40489					
✓ 39-32	60 PCS.	40489					

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date

1/2/78

REV

BY

CHKD

DATE

Reviewed (for material covered by code):

Authorized Inspector (Shop)

N/A

Date

18-1

Contract No.

13-6336

No.

Sh 1 of 39

MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

Piece-Mark	Serial No.	Material Heat No.	Matl. Type	Piece-Mark	Serial No.	Material Heat No.	Matl. Type
39 F	1 ASSY.	LEAK TEST ASSY.		39 R	1 ASSY.	LEAK TEST ASSY.	
✓ 39-11	2 PCS	40079		• 39-18	2 PCS	30384	
✓ 39-12	2 PCS	40079		✓ 39-2	2 PCS	40079	
39 G	1 ASSY.	LEAK TEST ASSY.		39 S	3 ASSY'S	LEAK TEST ASSY.	
✓ 39-13	2 PCS	40079		✓ 39-2	6 PCS	40079	
✓ 39-22	2 PCS	30384		• 39-19	6 PCS	30384	
39 H	2 ASSY'S	LEAK TEST ASSY.		39 T	2 ASSY'S	LEAK TEST ASSY.	
✓ 39-11	4 PCS	40079		✓ 39-2	4 PCS	40079	
✓ 39-14	4 PCS	40079		• 39-20	4 PCS	30384	
39 J	1 ASSY.	LEAK TEST ASSY.		39 W	1 ASSY.	LEAK TEST ASSY.	
✓ 39-13	2 PCS	40079		• 39-21	2 PCS	30384	
✓ 39-14	2 PCS	40079		✓ 39-12	2 PCS	40079	
39 K	1 ASSY'S	LEAK TEST ASSY.		39 Y	3 ASSY'S	LEAK TEST ASSY.	
✓ 39-2	12 PCS 2 PCS	40489 40079		✓ 39-13	6 PCS	40079	
✓ 39-15	14 PCS	40079		✓ 39-15	6 PCS	40079	
39 P	1 ASSY.	LEAK TEST ASSY.		39 A-A	1 ASSY.	LEAK TEST ASSY.	
• 39-16	2 PCS	30384		• 39-23	2 PCS	30384	
• 39-22	2 PCS	30384		✓ 39-2	2 PCS	40079	
39 M	2 ASSY'S	LEAK TEST ASSY.		39 B-B	1 ASSY.	LEAK TEST ASSY.	
✓ 39-13	4 PCS	40079		• 39-24	2 PCS	30384	
✓ 39-2	2 PCS	40079		✓ 39-25	2 PCS	40079	
✓ 39-1	2 PCS	40489		39 C-C	1 ASSY.	LEAK TEST ASSY.	
✓ 39-3	2 PCS	JR		✓ 39-26	2 PCS	40079	
✓ 39-4	2 PCS	KV		✓ 39-10	2 PCS	40489	

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date 11/30/75

 BY
CHKD
DATE

Reviewed (for material covered by code):

Authorized Inspector (Shop)

N/A

Date

118-1

Contract No.

13-6336

No. 39

Sh 2 of —

SUPPLEMENTAL SUMMARY SHEET
FOR MATERIAL VERIFICATION
CHICAGO BRIDGE & IRON COMPANY

GO 823

CONTRACT NO. <u>73-6336</u>		LOCATION <u>CCM</u>			
ITEM ID			ENGINEERING PIECE MARK	SERIAL NO.	NO. PIECES FABRICATED
30384	u m	5.6			
		13-5	39-16	1-4	4
			39-18	1-2	2
			39-19	1-6	6
			39-20	1-4	4
			39-21	1-2	2
			39-22	1-4	4
			39-23	1-2	2
		39-24	1-2	2	-
469494	u m	20-8	417-5	1-2	2
			422-7	1-4	4
			423-7	1-4	4
		218	365-4		1
			371-4		1
			377-4		1
			378-4		1
			386-4		1
		54-8	502-4	1-2	2
			520-7	1-2	2
			438-5	1-2	2
		63-8	302-5	1-10	10
			302-4	1-10	10
		17-8	473-8	1-2	2
		21-10	399-4	1-4	4
			342-5	1-2	2
			456-4	1-2	2
			400-4		1
		22-10	529-4		1
			434-4		1



Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 19 of 39 *rd/c*

GO 623 REV 11-72

CONTRACT NO. <u>73-6336</u>		LOCATION <u>CCM</u>				
Heat #	ITEM ID	Item #	NO. PIECES FABRICATED			
40079	5/06	38-6	19-3	39-2	101-1-8	28
				39-11	1-6	6
			20-3	39-25	1-2	2
				39-26	1-2	2
				39-27	1-54	54
				39-29	1-24	24
646111	2	42-11	43-4	81-3	2	1
				82-3	1	1
				91-1	1	1
				86-3	1-2	2
				85-3	1-2	2
				89-1	1	1
				80-1	1	1
				84-3	1	1
				82-1	1	1
				88-3	1	1
				85-1	1-2	2
				88-1	1	1
				86-1	1-3	3
				91-3	1	1
				90-3	1-6	6



MATERIAL HEAT
NUMBER SHEET

Material Types:

1. Welded Assemblies
2. Non-Welded Code Matl.
3. Non Code Matl.

Piece-Mark	Serial No.	Material Heat No.	Mett. Type	Piece-Mark	Serial No.	Material Heat No.	Mett. Type
39D-D	18 PCS	LEAK TEST ASSY					
✓ 39-28	18 PCS	40079					
✓ 39-27	50 PCS	40079					
✓ 39-3	18 PCS	JR					
✓ 39-4	18 PCS	KY					
39E-E	8 PCS	LEAK TEST ASSY					
✓ 39-30	8 PCS	40079					
✓ 39-29	18 PCS	40079					
✓ 39-3	8 PCS	JR					
✓ 39-4	8 PCS	KY					

Calculation L-001337
 Project No. 10248-012
 ATTACHMENT D
 Page 21 of 39 R/c

Data taken from records in accordance with applicable CB&I QA Manual.

CBI Shop QA

Date _____

1130/75—

Asy

BY

CHKD

DATE _____

Reviewed (for material covered by code):

Authorized Inspector (Shop)

N/A 18-7

Contract No.

3-6336

No. 39Sh 3 of

File Number 5.2 Document Title CMTB'S

File Number 5.2 Document Title CMTB'S

Contractor Name CB & I

Calculation L-001337

Project No. 10248-012

ATTACHMENT D

Page 22 of 39 R/c

Document Listing:

CMTA'S

Engineer Signoff _____ Paper Count 70

Paper Count 70

METAL GOODS

8200 Lehigh Avenue, Morton Grove, Ill. 60053
(312) 966-6000

Project No. 10248-012
ATTACHMENT D
Page 23 of 39 R/c

OUR P. O. # C-73-6336-112
IR NUMBER 24-91097
DATE SHIPPED 10/24/74
SPECIFICATIONS ASTMA-276
ITEM 304 Channel
3 x 1 3/8 x 3/16 x 20ft R/L
10/31/74

CERTIFIED TEST REPORT

Date: 10/21/74
HMH Register No. 005743
HMH Drawing No.
HMH Work Order No.

Description of Item

Weight and Pieces	Shape	Size	Length	Material
3,320 LBS 25 PCS	CHANNEL	3 x 1 3/8 x 3/16"	R/L	304 STAINLESS STEEL

SPEC: ASTM A 276

Chemical Analysis

Heat Number	C	S	P	CU	MO	CR	NI	SI	MN
40489	.051	.021	.030	.24	.09	18.56	9.96	.56	1.55

Mechanical Properties

Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
40489	89,200	51,600	50.0	68.0	

We hereby certify that the above figures are correct as contained in the records of this company.

By John D. Person Q.C. MANAGER
Name and Title

**ITT Harper Inc.**8200 Lehigh Avenue
(312) 966-6000

OUR NUMBER

24-90675

DATE SHIPPED

3-5-74

SPECIFICATIONS

ASTM-A-256

ITEM

① T304 SS Channel
2 x 1 x 3/16

CERTIFIED TEST REPORT

Date: 3/6/74

HMH Register No.

00430

HMH Drawing No.

S. Braun
4-5-7

HMH Work Order No.

Customer: METAL GOODS SERVICE CENTER
DIV. ALCAN ALUMINUM
11400 W. ADDISON STREET
Address: FRANKLIN PARK, ILLINOIS 60141Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 24 Of 39 R/O

Purchase Order No. 24-90675

Description of Item

Weight and Pieces	Shape	Size	Length	Material
3315 LBS 61 PCS	CHANNEL	2 x 1 x 3/16	19' 1/2 L	304 STAINLESS STEEL
920 LBS 11 PCS	CHANNEL	3 x 1 3/8 x 3/16	15-24 1/2 L	304 STAINLESS STEEL

Chemical Analysis

Heat Number	C	S	P	CU	MO	CR	NI	SI	MN	LBS
30660	.005	.025	.023	.30	.29	18.84	9.74	.57	1.41	2730
30549	.52	.015	.075	.23	.27	18.96	9.90	.46	1.60	585
40079	.011	.014	.021	.20	.19	18.55	9.63	.53	1.89	22

Mechanical Properties

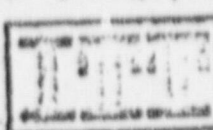
Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
30660	70,850	55,180	42	61.6	
30549	86,400	43,750	46.8	69	
40079	85,350	40,200	52.5	71	

We hereby certify that the above figures are correct as contained in the records of this company.

By

Name and Title

OC20



ITT Harper Inc.

8100 Leigh Avenue, Morton Grove, IL 60053
(312) 938-1000

* CORRECTION 10/17/73

CERTIFIED TEST REPORT

Date: 8/8/73

NMH Register No. 002751-C

Customer:

INDUSTRIAL SERVICE CENTER
255 BENT STREET
CAMBRIDGE, MASS. 02141

NMH Drawing No.

Address:

(NOW: Metal Goods Service Centers)

NMH Work Order No.

Purchase Order No. S-15536-D

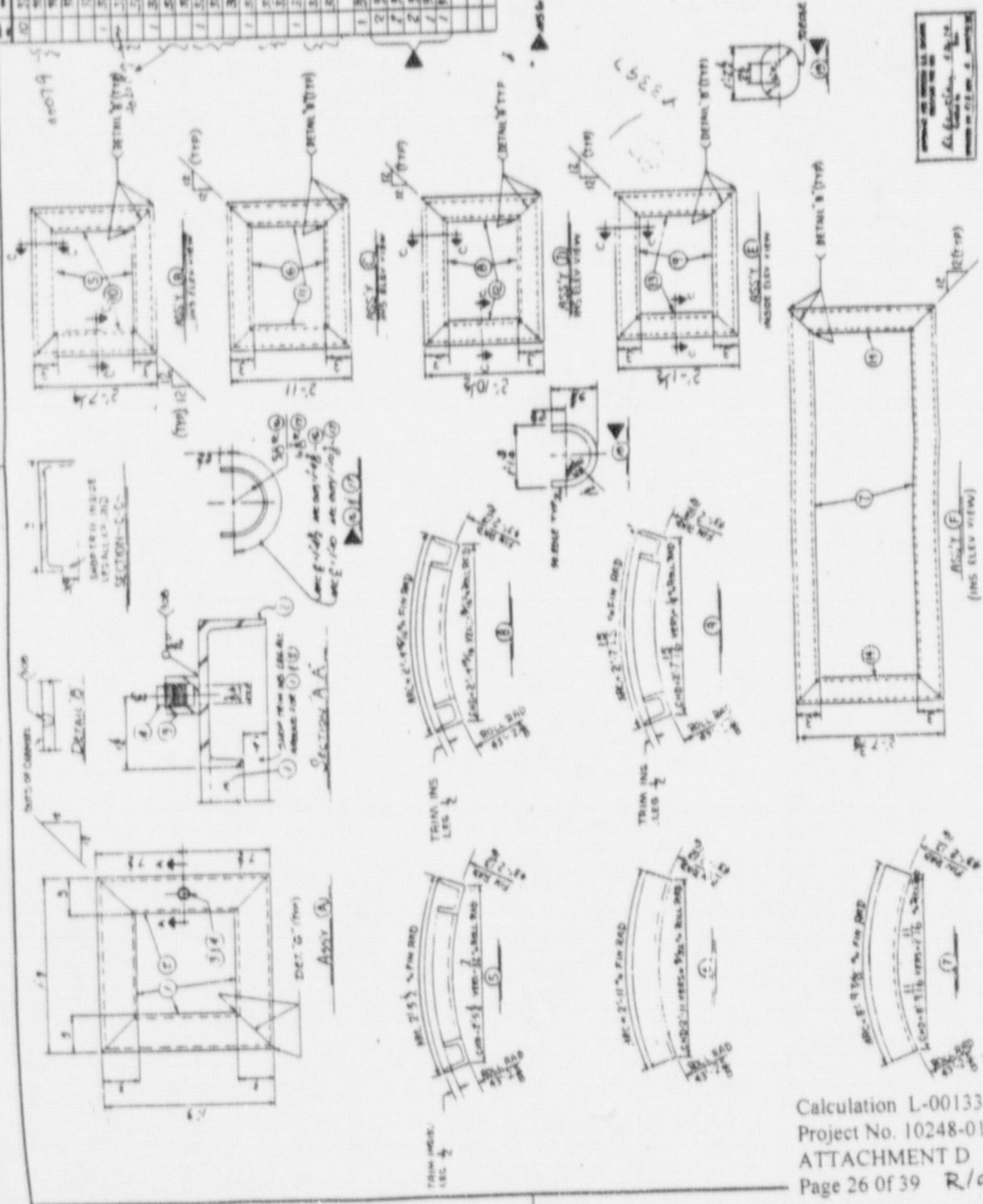
Description of Item				
Weight and Pieces	Shape	Size	Length	Material
3170 LBS 34 PCS	CHANNEL	3 x 1-3/8 x 3/16"	18/24'	304 STAINLESS STEEL

* SPEC ASME SA 479

Chemical Analysis										
Heat Number	C	S	P	CU	MO	CR	NI	SI	MN	CO
30408	.065	.020	.022	.32	.35	18.73	10.43	.45	1.52	
30384	.067	.021	.023	.37	.31	18.22	10.27	.37	1.83	

Mechanical Properties							
Heat Number	Tensile Strength	Yield Strength	%Elongation	R of A	Hardness	PCS	LBS
30408	92,000	52,200	50.8	71.5		20	1850
30384	91,400	51,500	51.3	71.7		14	1300

NY-10248-012



NO.	DESCRIPTION	QTY	UNIT	REMARKS
1	CONCRETE	10	CU YD	
2	STEEL	10	TON	
3	BRICK	10	CU YD	
4	CEMENT	10	TON	
5	SAND	10	CU YD	
6	GRAVEL	10	CU YD	
7	WATER	10	CU YD	
8	REINFORCING	10	TON	
9	PAINT	10	TON	
10	LABOR	10	MAN-HOUR	

RECEIVED
SARGENT & LUNDY
MAR 30 1978

STRUCTURAL DESIGN
GEOGRAPHIC DRAWING
1. SEE DRAWING FOR LOCATION OF BRIDGE
2. SEE DRAWING FOR GENERAL NOTES
3. SEE DRAWING FOR LOCATION OF BRIDGE
ON LEASE TEST AREA

Chicago Bridge & Iron Company
L. A. M. Test Assembly
NOT FOR CONSTRUCTION
77 6 33

CBI**MATERIAL HEAT
NUMBER SHEET**

Material Types:
 1. Welded Assemblies
 2. Non-Welded Code Matl.
 3. Non Code Matl.

Piece- Mark	Serial No.	Material Heat No.	Matl. Type	Piece- Mark	Serial No.	Material Heat No.	Matl. Type
32 B	1 ASSY.	LEAK TEST ASSY				Calculation L-001337	
✓ 32-5	2 PCS	40489				Project No. 10248-012	
✓ 32-10	2 PCS	40489				ATTACHMENT D	
						Page 28 of 39 R/c	
32 C	1 ASSY.	LEAK TEST ASSY					
✓ 32-6	2 PCS	40489					
✓ 32-11	2 PCS	40489					
32 D	1 ASSY.	LEAK TEST ASSY					
✓ 32-8	2 PCS	40489					
✓ 32-12	2 PCS	40489					
32 E	1 ASSY.	LEAK TEST ASSY					
✓ 32-9	2 PCS	40489					
✓ 32-13	2 PCS	40489					

Data taken from records in accordance with applicable CB&I QA Manual.
 CBI Shop QA 9/1/25 Date 1/30/25

REV BY
 CHKD
 DATE

Reviewed (for material covered by code):
 Authorized Inspector (Shop) N/A

116-1

Contract No.
73-6336

No. 32
 Sh 2 of —

Morton Grove, Ill. 60053

ITT Harper Inc.

8200 Lehigh Avenue
(312) 966-6000

OUR NUMBER 27-
DATE SHIPPED 3-5-74
SPECIFICATIONS ASTM-A-236
ITEM 1 T304 SS Channel
2x1x3/16
CERTIFIED TEST REPORT Date: 3/14/74

PORT
Date: 10/21/74
HMH Register No. 005743
HMH Drawing No.
HMH Work Order No.

CERTIFIED TEST REPORT
Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 30 of 39 R/0

HMH Register No. 001337
HMH Drawing No. 4-5-
HMH Work Order No.

Item	Length	Material
R/L		304 STAINLESS STEEL

Description of Item		
Size	Length	Material
3/16	19' 1/2 L	304 STAINLESS STEEL
2/16	14-24 R/L	304 STAINLESS STEEL

	NI	SI	MN
6	9.96	.56	1.55

Chemical Analysis							LBS
CU	MO	CR	NI	SI	MN		
.30	.29	18.84	9.27	.57	1.41		2730
.23	.27	18.96	9.90	.46	1.60		555
.20	.19	18.55	9.63	.53	1.83		22

ation	R of A	Hardness
	68.0	

Mechanical Properties			
Yield Strength	Elongation	R of A	Hardness
55,100	42	61.6	
43,750	46.8	69	
40,200	52.5	71	

ontained in the records of this

figures are correct as contained in the records of the

Person R.C. HANABER
Name and Title

By *[Signature]*
Name and Title

OUR P. U. #

IR NUMBER

DATE SHIPPED

SPECIFICATIONS

ITEM

(312) 966-8000

CERTIFIED TEST REPORT

Date: 10/21/74

HMH Register No. 005743

Calculation L-001337

Project No. 10248-012

ATTACHMENT D

Page 31 Of 39 R/c

HMH Drawing No.

HMH Work Order No.

131

Description of Item

Weight and
Pieces

Shape

Size

Length

Material

3,320 LBS 25 PCS	CHANNEL	3x1 3/8x3/16"	R/L	304 STAINLESS STEEL

SPEC: ASTM A 276

Chemical Analysis

Heat Number	C	S	P	CU	MO	CR	NI	SI	MN
40489	.051	.021	.030	.24	.09	18.56	9.96	.56	1.55

Mechanical Properties

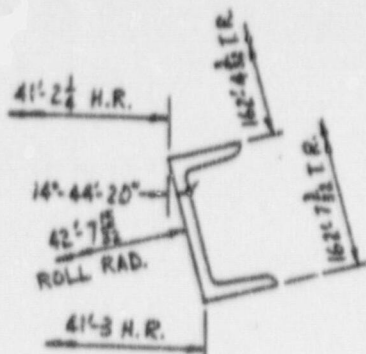
Heat Number	Tensile Strength	Yield Strength	Elongation	R of A	Hardness
40489	89,200	51,600	50.0	68.0	

We hereby certify that the above figures are correct as contained in the records of this company.

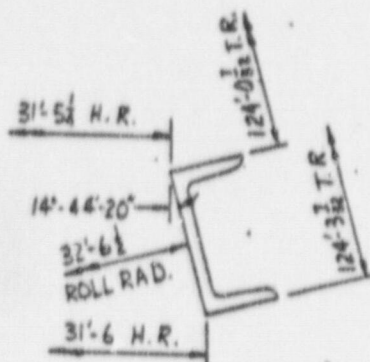
BY

Name and Title

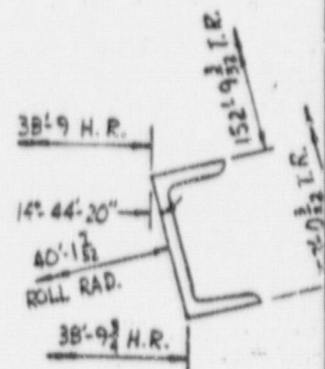
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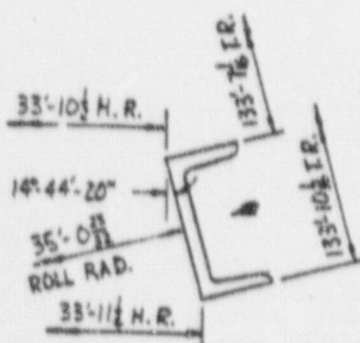
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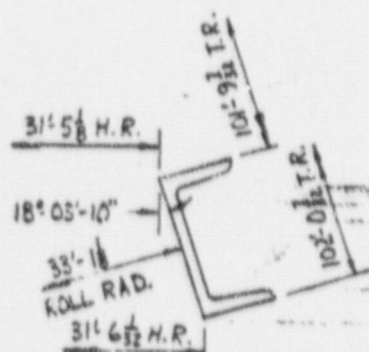
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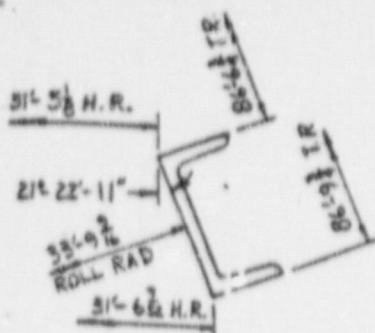
SECTION THRU PC ①



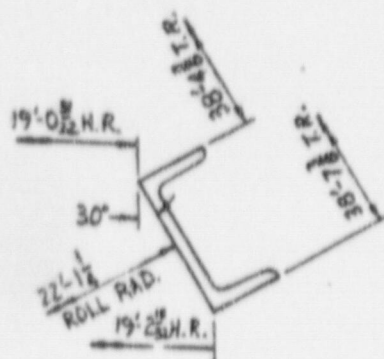
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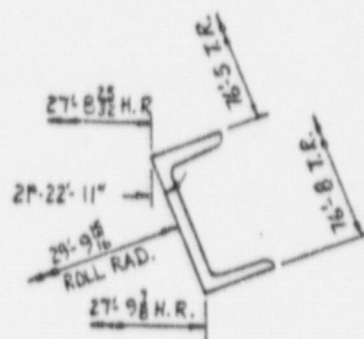
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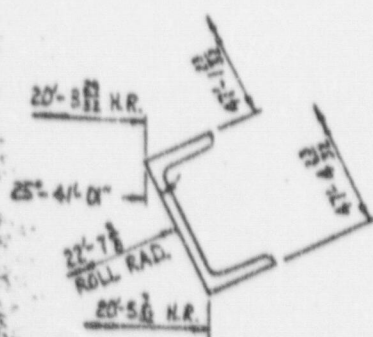
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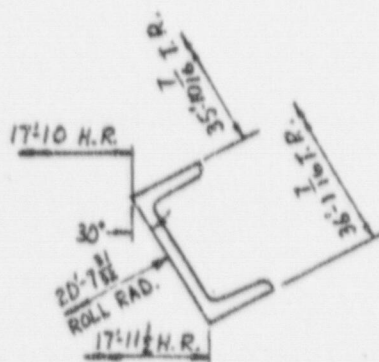
SECTION THRU PC (5)



SECTION THRU PC (11)



SECTION THRU PC (13)



SECTION THRU PC (14)

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PHOTOCOPYING, RECORDING, OR BY ANY INFORMATION STORAGE AND RETRIEVAL SYSTEM.



**SUPPLEMENTAL SUMMARY SHEET
FOR MATERIAL VERIFICATION**

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 34 Of 39 R/c

CONTRACT NO. <u>73-6-336</u>		LOCATION <u>C C M</u>			
SUPPLIER'S ID NO.		ENGINEERING PIECE MARK	SERIAL NO.	NO. PIECES FABRICATED	
D81784	100-4 2-12	603-2	1	1	
		611-3	1-2	2	
		611-4	1	1	
		609-2	1-4	4	
		609-3	1-2	2	
M92565	100-13 100-13 16-12	711-2		6	
		715-4		2	
		719-4		2	
		723-4		2	
94541	100-19 51-12	740-4	1-4	4	
		731-4	1-3	3	
		731-5		1	
		741-4	1-4	4	
KA4989	100-18 51-12	751-4		1	
		753-4		1	
		731-11	1-2	2	
		740-10	1-2	2	
		741-10	1-2	2	
94541	100-19 71-12	692-7	1-2	2	
		287-5	1	1	
		673-20	1	1	
		287-1		1	
		287-5		1	
KA4989 94541	100-18 82-12	287-11		1	
		455-5		1	
		293-12		1	
		293-13		1	
		293-14		1	
	110-12	283-2		1	
		283-5		1	



SUPPLEMENTAL SUMMARY SHEET
FOR MATERIAL VERIFICATION

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
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GO 823 REV 11-72

CONTRACT NO. <u>13-6336</u>		LOCATION <u>CCM</u>			
ITEM ID		ENGINEERING PIECE MARK	SERIAL NO.	NO. PIECES FABRICATED	
<u>94541</u>	<u>100-19</u>	<u>142-10</u>	<u>283-6</u>	<u>1</u>	<u>1</u>
			<u>283-3</u>	<u>1</u>	<u>1</u>
			<u>283-4</u>	<u>1</u>	<u>1</u>
	<u>94-1✓</u>		<u>287-12</u>		<u>1</u>
			<u>287-13</u>	<u>1</u>	<u>1</u>
			<u>287-14</u>		<u>1</u>
			<u>293-1</u>		<u>1</u>
			<u>293-4</u>	<u>✓</u>	<u>1</u>
			<u>293-5</u>		<u>1</u>
			<u>293-11</u>	<u>✓</u>	<u>1</u>
<u>801804750</u>	<u>100-12</u>	<u>115-12</u>	<u>712-2</u>		<u>48</u>
			<u>712-3</u>		<u>24</u>
			<u>716-1</u>		<u>16</u>
			<u>716-2</u>		<u>8</u>
			<u>720-1</u>		<u>16</u>
			<u>720-2</u>		<u>8</u>
			<u>724-1</u>		<u>16</u>
			<u>724-2</u>		<u>8</u>
			<u>717-3</u>		<u>6</u>
			<u>725-3</u>		<u>6</u>

DATE 80 OF 19



SUPPLEMENTAL SUMMARY SHEET
FOR MATERIAL VERIFICATION

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 36 Of 39 R/e

GO 823 REV 11-72

CONTRACT NO. 73-6336

LOCATION CCM

ITEM ID		ENGINEERING PIECE MARK	SERIAL NO.	NO. PIECES FABRICATED
✓ 88170	71-2	5-6		
		43-8	345-3	8
			379-3	8
			404-5	1
			404-4	1
			345-2	1
			875-2	1
		48-8	404-5	7
		82-8	343-2	1
			343-2	8
✓ 88170	71-2	8-10	426-3	8
			426-2	1
		107-10	526-2	1
		107-10	527-2	1
93063	62-4	130-12	526-3	8
			527-3	8
			287-13	1 new
			293-11 ✓	"
			293-1 ✓✓	"
			293-1 ✓	"
			293-5 ✓	"
			287-5 ✓	"
			287-11 ✓	"
			293-15 ✓	"
			287-15 ✓	"
			287-15 ✓	"
			287-15 ✓	"

SUPPLEMENTAL SUMMARY SHEET #12
FOR MATERIAL VERIFICATION

GO 823 REV 11-72

CONTRACT NO.		LOCATION				
Heat #	ITEM ID	Item #	S.G.#	ENGINEERING PIECE MARK	SERIAL NO.	NO. PIECES FABRICATED
40489	5106 ^{EE}	67-3	10-3	39-1	1-36	36
		In Barrent		39-30	1-8	8
				32-10	1-2	2
				32-13	1-2	2
				32-12	1-2	2
				32-11	1-2	2
				32-14	1-2	2
			15-3	39-31	1-60	60
				39-32	1-60	60
			19-3	39-2	1-100	100
				39-5	1-4	4
				39-6	1-2	2
				39-7	1-2	2
				39-8	1-12	12
				39-9	1-2	2
				39-10	1-4	4
			28-3	39-30	1	1
			55-3	32-5	1-2	2
				32-6	1-2	2
				32-7	1-2	2
				32-8	1-2	2
				32-9	1-2	2
			35-5	39-33	1	1
			32-5	39-35	1	1
			55-5	39-34	1	1
93063		67-4	117A	293-15 ✓		30

INLAND STEEL COMPANY

REPORT OF CHEMICAL AND PHYSICAL TESTS • METALLURGICAL DEPARTMENT



CONSIGNEE

DATE

CHICAGO BRIDGE & IRON CO.
P. O. BOX 774
KANKAKEE, IL. 60901

2 25 1975

DESTINATION

VIA

BILL ORDER NO.

SHIPPING NO.

INDIAN OAKS, IL

IC 97052

Z-43696

55380.4

5

SPECIFICATION & DESCRIPTION

ASTM A-36-70 C3X4.1#

100-19
Item #

PIECES	LENGTH		WEIGHT	CUSTOMER ORDER NO.	HEAT NO.	SLAB NO.	YIELD POINT LBS./SQ. IN.	TENSILE STRENGTH LBS./SQ. IN.	ELONGATION		BEND
	FT.	IN.							IN.	%	
201	30		24723	C73-6336G-159	94541		52200	72300	8	27	OK

CHEMICAL ANALYSIS

HEAT NO.	C	Mn	P	S	Si	Cu	Ni	Mo	Cr	Co	V	B	Al
94541	16	80	009	032									

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 38 of 39 R/e

STATE OF INDIANA

SUBSCRIBED AND SWORN TO BEFORE ME THIS

COUNTY OF LAKE

DAY OF

A.D. 19

NOTARY PUBLIC



WE HEREBY CERTIFY THAT THE ABOVE FIGURES ARE CORRECT
AS CONTAINED IN THE RECORDS OF THE COMPANY.

RAYMOND L. GILSON



JONES & LAUGHLIN STEEL SERVICE CENTER DIVISION
JONES & LAUGHLIN STEEL CORPORATION
CERTIFIED RECORD OF CHEMICAL AND MECHANICAL PROPERTIES



PURCHASER

Chicago Bridge & Iron Company
POST OFFICE BOX 744
Kankakee, Illinois 60701
Attn: Art Johnson

DATE 9-19-74

YOUR ORDER NO. C73-6336E-115 - C74-2270E10E
C73-6336F-12A

OUR INVOICE NO. 41-16082

CORRECTED

Calculation L-001337
Project No. 10248-012
ATTACHMENT D
Page 39 of 39/Final

MATERIAL AS DESCRIBED ON INVOICE

ITEM	MANUFACTURER	HEAT NO.	CARBON	MANG.	PHOS.	SULPHUR	SILICON	CHROM.	NICKEL	MOLY.	COPPER	T or Cl to
1	H.W.											
2	H.W.											
3	Inland											
1		60930	.24	.64	.018	.47						
2		60930	.24	.64	.018	.47						
3		93063	.14	.67	.008	.020	.21					
1		60930	66000	47300		25						
2		60930	66000	47300		25						
3		93063	68100	50400		27						

WE CERTIFY THE ABOVE INFORMATION IS AN EXACT COPY OF CERTIFIED TEST CERTIFICATES FURNISHED BY THE ABOVE MANUFACTURERS AND KEPT IN OUR PERMANENT RECORD FILES.

OF September 19 74

R. McCullough
NOTARY PUBLIC

MY COMMISSION EXPIRES 1-22-1975

SIGNED

CHICAGO

William Carl

SERVICE CENTER

JONES & LAUGHLIN STEEL SERVICE CENTER DIVISION

Attachment K

References

References

1. NRC Inspection Report 50-373/78-33, dated January 19, 1979
2. L. O. DelGeorge (ComEd) letter to A. Schwencer (NRC), "LaSalle County Station, Units 1 and 2, Containment Leak Chase Channels," dated August 10, 1981
3. L. O. DelGeorge (ComEd) letter to A. Schwencer (NRC), "LaSalle County Station, Units 1 and 2, Supplemental Information Concerning Containment Leak Chase Channels," dated September 1, 1981
4. L. O. DelGeorge (ComEd) letter to A. Schwencer (NRC), "LaSalle County Station, Units 1 and 2, Containment Leak Chase Channels," dated October 1, 1981
5. NRC Inspection Report 50-373/81-28, dated November 6, 1981
6. S. V. Athavale (NRC) letter to Donald F. Schnell (Union Electric Company), "Containment Liner Leak Chase Channel Venting Callaway Plant Unit No. 1 (TAC No. 72750)," dated March 14, 1990
7. Stephen P. Sands (NRC) letter to Thomas J. Kovach (ComEd), "Safety Evaluation of Containment Leak Chase Channels - Byron, Unit Nos. 1 and 2, Braidwood, Unit Nos. 1 and 2 (TAC Nos. 72569, 73570, 66476, and 66477)," dated May 17, 1990
8. Licensee Event Report (LER) 50-373/97-030-00, dated September 17, 1997