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# Evaluation of Materials of Construction for the Reinforced Concrete Reactor Containment Model

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Prepared by G. A. Knorovsky, P. W. Hatch, M. R. Gutierrez

Sandia National Laboratories

Prepared for  
U.S. Nuclear Regulatory  
Commission

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Prepared by  
G. A. Knorovsky, P. W. Hatch, M. R. Gutierrez

Sandia National Laboratories  
Albuquerque, NM 87185

Prepared for  
Division of Engineering  
Office of Nuclear Regulatory Research  
U.S. Nuclear Regulatory Commission  
Washington, DC 20555  
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## Abstract

This report summarizes the chemical analysis, metallography, and tensile test results obtained from steel materials and welds used in the construction of a 1/6-scale model reinforced concrete nuclear reactor containment. The purpose of building such a model is to experimentally verify the ability of numerical models to predict deformation and failure in full-size containments. Naturally, the predictions of such models are strongly influenced by the constitutive models used for the containment materials. The program reported here was intended to provide such data. Besides providing tensile test data on the sheet, plate, and rebar materials used for dome, cylinder, cylinder inserts, reinforcements, and penetrations, pull tests on several weld geometries have been obtained. Standard tensile test data derived from load vs. extensometer output records (engineering stress vs engineering strain) are supplemented with true stress vs true strain data collected by measuring actual cross sectional areas during interrupted tensile tests. Additionally, "r" value measurements (ratio of width to thickness strain) which can be used to provide information about anisotropic multiaxial flow conditions are derived from the true strain data and reported. A small number of tests that included holds at constant load levels were also performed and analyzed to yield ambient temperature creep equations. Finally, conditions resulting from construction practices (such as residual stresses and the Bauschinger effect) that may affect the prediction of yield and fracture stresses from this data are briefly discussed.

## Table of Contents

Introduction	1
Chemical Analysis	1
Tensile Test Results	2
Metallographic Results	4
Discussion	4
Elastic Properties	4
Multiaxial Yield and Failure Criteria	4
Variation of Weld Properties	6
Model Containment Test Procedure vs Tensile Testing Procedure	6
Summary	7
Bibliography	8
Distribution	46

<u>Figure</u>	<u>List of Figures</u>	<u>Page</u>
1	Engineering stress/strain curve for dome material	15
2	Engineering stress/strain curve for cylinder material	15
3	Engineering stress/strain curve for penetration sleeve material	16
4	Engineering stress/strain curve for insert plate material	16
5	True stress/strain data for dome material a) pre-dished condition, b) post-dished condition	17
6	True stress/strain data for cylinder material	18
7	True stress/strain data for insert plate material	18
8	True stress/strain data for penetration sleeve material	19
9	Additional weld specimen geometries: a) Penetration sleeve (1.3" reduced to 0.5", to insert plate (0.2")), with backup bar, b) Equipment hatch dome (0.2") to tension ring, c) Bottom corner to 0.07" bottom with 1" x 2" backing bar (does not illustrate 0.07" cylinder to bottom corner weld), d) 0.07" bottom to 0.07" bottom with 1" x 2" backing bar, e) 0.07" cylinder to bottom corner, f) Stud welds, 1/8" shaft to plates 0.07" thick, 0.09" thick and 0.2" thick g) Details of loading geometries	20 21
10	a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/cylinder specimen	22

<u>Figure</u>	List of Figures (cont'd)	<u>Page</u>
11	a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/insert plate specimen	23
12	a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/dome specimen	24
13	a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for dome/dome specimen	25
14	Cross-weld load/elongation curve for cylinder/bottom corner specimen	26
15	Cross-weld load/elongation curve for penetration sleeve/insert plate specimen	26
16	Cross-weld load/elongation curve for equipment hatch dome to tension ring specimen	27
17	Load/elongation curve for stud to cylinder weld	27
18	Cross-weld load/elongation curve for bottom corner to bottom with backing bar	28
19	6 mm rebar load/engineering strain curve	29
20	#3 rebar load/engineering strain curve	29
21	#4 rebar load/engineering strain curve	30
22	#5 rebar load/engineering strain curve	30
23	True width and thickness strain vs. true longitudinal strain for dome material	31
24	True width and thickness strain vs. true longitudinal strain for cylinder material	31
25	True width and thickness strain vs. true longitudinal strain for penetration sleeve material	32

List of Figures (cont'd)

<u>Figure</u>		<u>Page</u>
26	True width and thickness strain vs. true longitudinal strain for insert plate material	32
27	Cross-section of stud weld to cylinder material (8.5x)	33
28	Cross-section of stud weld to insert plate material (8.5x)	33
29	Cross-section of weld between dome sections (flat position 10x)	34
30	Cross-section of weld between dome sections (vertical position 12x)	34
31	Cross-section of weld between dome and cylinder (horizontal position 8.5x)	35
32	Cross-section of weld between cylinder sections (horizontal position 10x)	35
33	Cross-section of weld between cylinder sections (vertical position 10x)	36
34	Cross-section of weld between cylinder sections (flat position 10x)	36
35	Cross-section of weld between cylinder and insert plate (horizontal position 13x)	37
36	Cross-section of weld between cylinder and insert plate (vertical position 12x)	37
37	a) Forming limit curves for nine grades and lots of rimmed, low-carbon steels, b) Forming limit curves for six grades and lots of aluminum-killed (AK), low-carbon steels after Hecker (ref. 3)	38
38	Comparison of monotonic and programmed load tensile behavior for cylinder material	39
39	Comparison of monotonic and programmed load tensile behavior for dome material	39
40	Comparison of monotonic and programmed load tensile behavior for insert plate material	40



<u>Figure</u>	List of Figures (cont'd)	<u>Page</u>
41	Comparison of monotonic and programmed load tensile behavior for penetration sleeve material	40
42	Comparison of monotonic and programmed load tensile behavior for #4 rebar	41
43	Creep behavior of cylinder material	42
44	Creep behavior of dome material	42
45	Creep behavior of insert plate material	43
46	Creep behavior of penetration sleeve material	43
47	Creep behavior of #4 rebar	44
48	Compilation of creep equation coefficients: a) Coefficient A, b) Coefficient B	45

List of Tables

<u>Table</u>		<u>Page</u>
I	Chemical Compositions	9
II	Sheet and Plate Tensile Properties	10
III	Weld Tensile Properties	12
IV	Rebar Tensile Properties	14

## Introduction

As part of Sandia National Laboratories' Reactor Containment Integrity Program, an approximately 1/6th scale model of a generic reinforced concrete reactor containment building was designed, built, comprehensively instrumented, and tested to failure by internal pressurization. Materials and construction practices were intended to duplicate those used in full-size containments as much as possible. The purpose behind this effort is to study the behavior of such structures under beyond-design-basis conditions, and furthermore to verify the capability of existing computer codes to predict such behavior. Because the predictions of such computer codes are dependent upon accurate material mechanical response models, it was necessary to measure mechanical properties for the materials of construction. This report is intended to document the extensive data gathering effort which was conducted in response to this need, and as such will be a supplement to the overall documentation package reporting on the program. Examples of every material or weld in the containment were not tested; however a broad range representative of most of the steel construction was evaluated. Concrete material properties are covered in a separate report [1]. Included in the body of this report are standard engineering stress/strain data and representative curves, supplemental true stress/strain data obtained to strains beyond the maximum load, derived values of the ratio between true width and thickness strains (the "r" value), ambient temperature creep data, chemical analyses, descriptions of how the data were obtained, and finally discussion on how the uniaxial data measured can be applied to the multiaxial stress and strain condition of the actual containment.

## Chemical Analyses

Emission spectroscopy results obtained from the sheet and plate materials and reinforcing bars ("rebar") are summarized in Table I (note that the materials are defined by their thicknesses or diameters in the tables which follow; their locations in the containment model are defined below). Specifications under which the materials were purchased include: (1) liner (0.07" cylinder or bottom, and 0.09" dome materials): ASME SA-414 Grade D, (2) 0.2" liner insert: ASME SA-516 Grade 60, (3) large penetration sleeves (1.3" reduced to 0.5"): ASME SA-516 Grade 60, and (4) rebar: ASTM A-615. Supplemental requirements instituted by the constructor on the SA-414 liner material required that it be fully killed and made to fine grain practice. Analysis of the chemistries shown in Table I indicates that the A-414 materials were somewhat unusual in that they were killed with Al only, instead of the more usual Si or Si plus Al practice. Of all

the materials tested, only the dome material did not exhibit a sharp yield point. Interestingly, the cylinder material, with nearly identical composition, did exhibit a sharp yield point. With respect to meeting specification requirements, all the materials except for the 0.2" SA-516 are within specification. The SA-516 exceeds the allowed maximum limitation on Mn of 0.90 wt.%. Possibly this material was procured in larger thickness and reduced to its 0.2" value; the Mn specification limit increases to 1.20 wt.% for material exceeding 0.5". The rebar all appears to be A-615 grade 60 material. The 6mm bar was manufactured to an European standard, though it appears to meet all A-615 grade 60 chemical and tensile requirements.

### Tensile Test Results

Standard sheet tensile specimens of 0.5" gauge width by 2" gauge length by material thickness (or 0.25", whichever was less) were machined from sheet and plate materials provided. Only the material taken from 1.3" thick plate had all four sides of the gauge length machined; the others taken from sheet were machined on the edges only, with the as-rolled surfaces left as received (oxidized from atmospheric exposure). In addition to pre-dished dome material (0.09"), post-dished (cold worked) dome material was also tested. Tensile specimens were taken in both longitudinal and transverse directions with respect to the rolling direction. At least two specimens were pulled for each combination of material and orientation tested. All specimens were tested at ambient temperature. The nominal strain rate was  $10^{-3}$  sec.<sup>-1</sup>. Engineering stress/strain data for the tensile specimens are summarized in Table II, and representative curves for each material are given in Figures 1-4. It should be noted that the small values of the standard deviation given in Table II for several of the materials may not be representative of the variation that would actually be expected with a greater number of specimens. It would be prudent to choose a pooled value, which for the yield and ultimate stresses are: 0.52 and 0.42 ksi respectively.

True stress/strain data are summarized in Figures 5-8 and Table II. The true stress/strain data were determined by stopping the tensile test at intervals and measuring the actual cross sectional area at five locations along the gauge length using dial calipers. These measurements are taken to beyond the point at which diffuse necking begins (past the maximum load condition), and so provide true strain data to higher strains than are usually obtainable by conversion from engineering stress/strain curves. Until local necking occurs near failure (such data would deviate from the straight line through data at lower strains on the ln true stress/ln true strain plot), this enables good measurement of stresses and strains. A Bridgman-type correction [2] was not applied to compensate for the effect of non-axial stress components. It

was noted that the actual strains measured (given by  $\epsilon = \ln[A_0/A]$ , where  $A$  = instantaneous area and  $A_0$  = original specimen cross sectional area) were significantly different from the average strains measured by the extensometer even at the smallest true strains (~5%) measured.

Sheet metal butt welds were tested in the cross-weld orientation (for which load data, and not stress, is reported), and in the longitudinal (all-weld-metal) orientation. The cross-weld specimens were not machined on all four sides, thus the weld reinforcement prevented a uniform cross section, and failure always took place outside the weld region. The all weld metal specimens were fully machined to a gauge cross section of ~0.25" wide by ~0.05-~0.08" thick depending on the weld size. In addition to the sheet metal butt seam welds, samples of various other weld geometries were made. These specimens were nominally 1" wide, and were pulled to failure. If the failure was not in the weld fusion or heat affected zone, this region was machined down to a reduced width, and another sample pulled, at which time failure did occur in the weld region. Details of these geometries, and the manner in which they were loaded are given in Figure 9. Maximum loads for both the full and reduced width samples are given in Table III, as is the tensile data for the all weld metal specimens and the stud weld specimens. Representative load-strain or load-elongation cross-weld and engineering stress-strain all weld metal curves are shown in Figures 10 - 18.

As in the case of the cross-weld samples, load vs. extension data only are provided for the rebar materials, which were not machined to a uniform section, but were tested as received. Load data are given in Table IV, and representative load vs engineering strain curves are given in Figures 19-22.

"r" value data is summarized in Table II (refer to the discussion section for a definition of "r"). Linear curves were fit to data plots of true width and thickness strain vs true longitudinal strain as shown in Figures 23-26 (note that the "R" value given on these figures refers to the goodness of fit of the straight line through the data points). The first of the two "r" values given in Table II refers to a value obtained at the maximum load as suggested in ASTM E-517 (obtained by inserting the longitudinal true strain at maximum load in the linear best-fit equations for true width and thickness strain, and taking the ratio of these values), and the second by taking the ratio of the slopes of the best-fit curves. Because the straight lines did not in general go exactly through the origin, the two values differ. Where the values differ, the authors prefer the ratio of the slopes value, as the straight lines should go through the origin. Small systematic measurement errors are reflected more in the intercept than in the slope; furthermore, the most accurate

data occurs at larger strains but before local necking takes place, simply because of the precision of the calipers used for area measurement.

### Metallographic Results

Welds were sectioned metallographically to determine if any porosity or other defects were present. With the exception of the stud welds to the 0.2" material, no serious problems were noted. These latter stud welds exhibited poor consistency, with extensive porosity and cracking in some of the welds, while others broke the shaft of the stud during tensile testing (at ~1600 lbs.). These conditions are shown in Figures 27 & 28. The load-extension data (see Table III) also exhibited large scatter, and several of the welds were broken off in handling. Stud welds to the thinner sheet showed consistently good strength, with only minor porosity being present. Examples of the other sheet metal welds are shown in Figures 29-36.

### Discussion

Elastic Properties-Previous data gathering for reactor containment model testing [3] also included determining values of the elastic moduli and Poisson's ratio. Since values obtained in the earlier work showed little variation from handbook values, such tests were not conducted.

Multiaxial Yield and Failure Criteria-The standard tensile test data summarized in this report provide information on the behavior of materials under uniaxial stress conditions. However, as the testing of the model containment is under multi-axial conditions, yield and failure criteria for multiaxial conditions must be chosen in order to compare computer code predictions with experiment. A common choice for the yield condition is the Von Mises criterion. This assumes isotropic behavior of the material; however, it may be extended to anisotropic materials if certain material constants are known (via Hill's analysis, as explained in [4]). The complexity of the mathematical equations which describe the yield locus and hence, required number of material constants, increases depending upon upon how anisotropic the material is. For the special case of planar isotropy (an important case for sheet material, which corresponds to equal uniaxial yield stresses in the rolling and transverse directions, but not in the thickness direction) only one constant needs to be determined. This constant, the plastic strain ratio or "r" value, is the ratio of strain in the transverse direction to the tensile axis, i.e. the width strain, to the thickness strain, with the ratio averaged over specimens taken parallel to the rolling direction, transverse direction and 45 degrees to rolling direction. It relates the ratio of the yield stress in the thickness direction:  $\sigma_3$  to

that in plane:  $\sigma_1$  (the yield stress in the rolling direction) or  $\sigma_2$  (the yield stress in the transverse direction) by the equation:

$$\sigma_3/\sigma_1 = \sigma_3/\sigma_2 = [(1 + r)/2]^{1/2}$$

The usefulness of this relation becomes evident when it is realized that the yield stress under balanced biaxial tensile loading is equivalent to the compressive yield stress in the thickness direction (the two stress states differ only by a hydrostatic compressive stress equal to the yield stress in the thickness direction, and since the yield locus is assumed insensitive to hydrostatic pressure, the values are equivalent). An approximate "r" value can be obtained from the interrupted true stress/strain test data obtained in this program (no tests were conducted in the 45 degree direction as is required in the ASTM E-517 Standard). Values of this parameter for the various materials of construction are tabulated in Table II. Within the accuracy of the measurements no significant difference in "r" between the rolling and transverse directions was noted. In agreement with this the values of yield stress for the rolling and transverse directions in Table II can also be seen to be insignificantly different. Hence the planar isotropy condition appears to hold reasonably well. The values of r obtained are significantly different from 1.0 in nearly all cases; however, they are not so different that an analysis based upon isotropic behavior would be seriously affected.

In addition to these considerations for predicting biaxial yielding, two other influences need to be considered. The first is that any welded vessel will contain substantial residual stresses of both tensile and compressive nature. Secondly, since the deformation of the sheet and plate materials necessary to shape them into their final cylindrical or spherical forms was accomplished without subsequent annealing, the Bauschinger effect must be taken into account. Both of these influences will be expected to wane as a few percent deformation occurs, and they will not influence fracture (assuming that ductile fracture occurs).

Beyond the yield condition, empirically-determined Formability Limit Curves are more useful for predicting failure under biaxial loading conditions. Data for sheet steels of similar grades to that tested has been measured by Hecker [5]. An example of his data is shown in Fig 37. As may be seen in Fig. 37, the minimum condition of strain is usually near to the plane strain condition ( $\epsilon_2 = 0$ ;  $\epsilon_1$  &  $\epsilon_3 \neq 0$ ). Formability limits found in this manner assume that proportional straining is occurring; changing the strain path from other than radially outward from the origin can invalidate the limits given. Fortunately, for non-proportional straining where the

loading increases monotonically the limits tend to be conservative. Other investigators [6] have found that the ratio of the true plane strain ductility to the true uniaxial strain ductility is ~0.7 for a large range of steels of varying strengths. The maximum uniaxial true strains measured by the interrupted straining technique are summarized in Table II.

Determining a triaxial failure criterion analogous to the biaxial Forming Limit Curves is a non-trivial task. It is known that ductility is greatly reduced as the ratio of hydrostatic tensile to deviatoric stresses is increased [7]. Since failure of the test vessel will probably be found due to fracture initiation at a flaw or structural discontinuity anyway, it is questionable if a triaxial failure criterion for homogeneous material is really useful. Thus, it would probably be more appropriate to apply fracture mechanics considerations. No fracture mechanical testing was included in this program, though material has been archived should it become desirable in the future.

Variation of Weld Properties-The data for sheet weld metal as presented in Table III is divided into categories of size and weld position. Category choices were made by performing an analysis of variance on the data for yield stress (or load, for cross-weld specimens) and ultimate tensile stress (or maximum load) versus thickness and position of weld. Those found significantly different were separated and reported individually; otherwise data was grouped. At least four weld processes (Shielded Metal Arc, Gas Metal Arc, Gas Tungsten Arc and Capacitor Discharge Stud Welding) were specified in the construction of the vessel.

Model Containment Test Procedure vs Tensile Testing Procedure-The standard tensile test procedure is performed under monotonic loading conditions, whereas the model test involves a series of load increases followed by varying length hold periods. Since it is known that carbon and nitrogen can diffuse at room temperatures, it was felt desirable to check for evidence of strain ageing in the construction materials, and the effect it might have on the stress-strain relations. Two types of tests provided this data. First, the interrupted straining tests which gathered true strain data provided qualitative information on whether a load jump occurred upon reloading. However, since these were not optimum for gathering quantitative data, and none were performed on the rebar, a few computer-controlled tensile tests in which 4, 5 or 10 ksi (in engineering stress) load increments followed by constant load ageing steps (for 30 minutes) were also performed. The interrupted straining tests showed that only the Si/Al killed materials seemed to show evidence of ageing during the hold intervals. However, the load controlled tests did not seem to exhibit any such behavior. Instead, these



tests showed that after the creep strain interval the flow stress increased abruptly to the level which would be expected at the strain accumulated regardless of whether it was accumulated during monotonic loading or as a consequence of the interrupted loading path. This behavior is demonstrated in Figures 38-42 (the slight discontinuities seen in the monotonic loading specimens are an artifact of changing extensometer strain sensitivity during the test). The engineering strains accumulated during the constant load hold periods are plotted as log/log plots in Figures 43-47 as a function of engineering stress level and time. The strain data seem to be well-represented by a simple equation of the form:

$$\log \epsilon = A + B \log t$$

The coefficients A and B are plotted as a function of stress level for each of the five major material types in Figure 48. Readers are cautioned that extrapolation of these equations beyond the stress, strain and time limits given by the data endpoints in Figures 43-47 may be questionable, as they are only obtained from curve-fitting and are not model-based. It was noteworthy that the amount of creep strain increased rapidly as the ultimate load was approached, and several specimens actually failed during their creep load holds. A value of the failure strain was not obtained as it exceeded the strain shut off limit of the software controlling the test (14%).

#### Summary

This report summarizes the chemistries and mechanical properties of materials of construction (including welds) for the 1/6-scale model reinforced-concrete reactor containment model tested in July of 1987 by Sandia National Laboratories' Containment Integrity Division. Measured properties reported included: 1) engineering yield and ultimate stresses and strains determined from monotonic tensile tests, 2) true stress/strain values to beyond maximum load determined from interrupted load tensile tests, and 3) creep strain accumulated during special programmed loading tests. Further, a measure of the yield surface anisotropy, the "r" value, or plastic strain ratio was also determined approximately. Additional comments concerning the materials of construction, the quality of welds, and the effect of other phenomena on the yield behavior were also noted. Should further testing be necessary, quantities of the materials will be archived for a reasonable length of time.

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Table I - Chemical Compositions

thickness or nom. dia.:	<-----sheet/plate----->					<-----rebar----->		
	0.07"	0.09"	0.2"	1.3"	6mm	No. 3 3/8"	No. 4 4/8"	No. 5 5/8"
C	0.17	0.15	0.21	0.16	0.21	0.44	0.46	0.43
Mn	0.73	0.70	1.04	1.21	1.14	0.71	0.93	1.01
Si	<0.01	<0.01	0.18	0.23	0.46	0.15	0.19	0.15
S	0.009	0.006	0.021	0.015	0.018	0.041	0.03	0.039
P	0.015	0.018	0.011	0.011	0.027	0.022	0.01	0.024
Ni	<0.01	<0.01	<0.01	0.20	0.03	0.11	0.07	0.08
Cr	0.02	0.02	0.02	0.20	0.02	0.16	0.10	0.10
Mo	<0.01	<0.01	<0.01	0.05	<0.01	0.03	0.01	0.02
Cu	<0.01	0.02	<0.01	0.19	0.02	0.23	0.23	0.27
Ti	<0.01	<0.01	<0.01	<0.01	<0.01	<0.01	<0.01	<0.01
V	ND	ND	ND	ND	<0.01	ND	ND	<0.01
Al	0.04	0.04	0.03	0.02	<0.01	<0.01	ND	ND

All values in wt.%; ND means not detected. Other elements analyzed for included: B, Co, Nb, Sn, Ta, W, & Zr; they were either ND or only barely detectable (<0.01).

Table II - Sheet and Plate Tensile Properties

Material	yield stress	ultimate tensile strength	elongation at:	
			max. load	failure
1.3"R.D.#	51.7 ksi (0.8)*	76.1 ksi (0.2)*	16% (0.2)*	45 % (1)*
1.3"T.D.#	51.6 (0.4)	75.2 (0.4)	16 (0.2)	41 (0.5)
0.2"R.D.	50.7 (0.4)	76.6 (0.7)	15 (2)	41 (1)
0.2"T.D.	51.4 (0.1)	76.9 (0.4)	15 (1)	32 (1)
0.09"T.D. pre-dished	53.0 (0.4)	71.2 (0.2)	15 (4)	23 (6)
0.09"R.D. Pre-dished	52.1 (0.6)	71.0 (0.6)	16 (1)	37 (3)
0.09"T.D.	52.4 (0.1)	69.6 (0.1)	16 (1)	25 (3)
0.09"R.D.	50.7 (0.5)	69.8 (0.2)	18 (0)	38 (0)
0.07"T.D.	49.4 (0.1)	70.6 (0.4)	16 (0)	34 (8)
0.07"R.D.	50.2 (1.0)	69.7 (0.4)	17 (0)	26 (6)

# R.D.= rolling direction; T.D.= transverse direction.

\* Standard deviation.

Two specimens each were tested for the 0.07 & 0.09" materials; three each were tested for all others.

Table II(cont'd) - Sheet and Plate Tensile Properties

True Stress/True Strain Data

$$\ln \sigma = A + B \ln \epsilon$$

where  $\sigma, \epsilon$  are true stress and true strain, respectively.

Material	A		B		Fracture Strain		r&	r@
0.07"	3.89	(0.08)*	0.19	(0.02)*	0.73	(0.10)#	1.00	0.74
0.09" pre-dished	3.84	(0.12)	0.20	(0.03)	0.89	(0.11)	0.82	0.77
0.09"	3.94	(0.03)	0.18	(0.01)	0.83	(0.065)	0.81	0.71
0.2"	4.04	(0.02)	0.17	(0.01)	0.71	(0.19)	0.93	0.93
1.3"	3.95	(0.05)	0.20	(0.01)	1.10	(0.14)	1.26	1.08

\*90% confidence limit deviation

#standard deviation

& = "r" value at max. load

@ = "r" value from ratio of slopes

Table III - Weld Tensile Properties

Weld Identification	Transverse Strength		Fracture Location
	Yield	Maximum	
0.07" to 0.07" with 1" x 2" backing bar	3370 lbs/in	4700 lbs/in	base metal
	(40)#	(15)#	
	3540 (360)	5260 (670)	*fusion line
0.07" to bottom corner	3590	4860	base metal
	(20)	(20)	
	3100 (70)	5140 (50)	*fusion line
Penetration (1.3" mat'l reduced to 0.5") to 0.2" insert plate	10300	15300	base metal
	(60)	(100)	
	11100 (140)	18400 (420)	*fusion line
Penetration dome (0.2") to Penetration (1.3" mat'l reduced to 0.5")	--	11000 (3550)	2 at fusion line, 1 in base metal
Bottom corner to 0.07" bottom with 1" x 2" backing bar	3740	4920	base metal
	(730)	(120)	
	3980 (40)	5440 (250)	*fusion line
Stud weld to 0.07" sheet	--	926 (110)	pulled nuggets
Stud weld to 0.09" sheet	--	1245 (375)	"
Stud weld to 0.2" sheet	--	1125 (639)	2 broke in stud, 2 pulled small nuggets, 2 broke off in handling (not counted for average and sd).

#Standard deviation

\*Reduced in width at fusion zone to force failure into weld.

Table III - Weld Tensile Properties (cont'd)

0.07" sheet to 0.07", 0.09" sheet or 0.2" sheet	3360 lb/in (140)	4590 lb/in (80)	base metal 18% (sd 1%) elongation to failure in 2" gauge length for 0.07" mat'l.
0.09" to 0.09" sheet	4280 lb/in (120)	5840 lb/in (40)	base metal 16% (sd 2%) elongation to failure in 2" gauge length.

All Weld Metal Longitudinal Tensile Properties

	yield stress	ultimate tensile strength	elongation at: max load	failure
0.07" to 0.07" 0.09" to 0.09" flat and vertical welding positions	67.6 ksi (2.6)	84.9 ksi (3.7)	8% (1)	11% (2)
0.07" to 0.07" 0.07" to 0.09" horizontal welding position.	63.8 (1.8)	79.8 (3.0)	8 (0)	10 (0)
0.07" to 0.2" vertical welding position.	85.5 (2.9)	104.7 (5.3)	*	*
0.07" to 0.2" horizontal welding position.	74.8 (3.9)	95.9 (4.4)	5 +	7 +

\*All samples broke outside extensometer gauge length.

+Only one sample did not break outside extensometer gauge length.

Table IV - Rebar Tensile Properties

Material	Yield Load	Maximum Load	Elongation at:	
			Max. load	Failure
6 mm	3340 lb (60) #	4330 (70) #	10% (2) #	* *
No.3	7160 (130)	11020 (190)	11 (1)	* *
No.4	12730 (280)	20440 (380)	11 (1)	21% (2)
No.5	21240 (310)	32850 (300)	11 (1)	22 (0)

# Standard deviation.

\*All specimens broke outside the extensometer gauge length.  
Four specimens were tested for each size except No.  
4, where 20 were tested



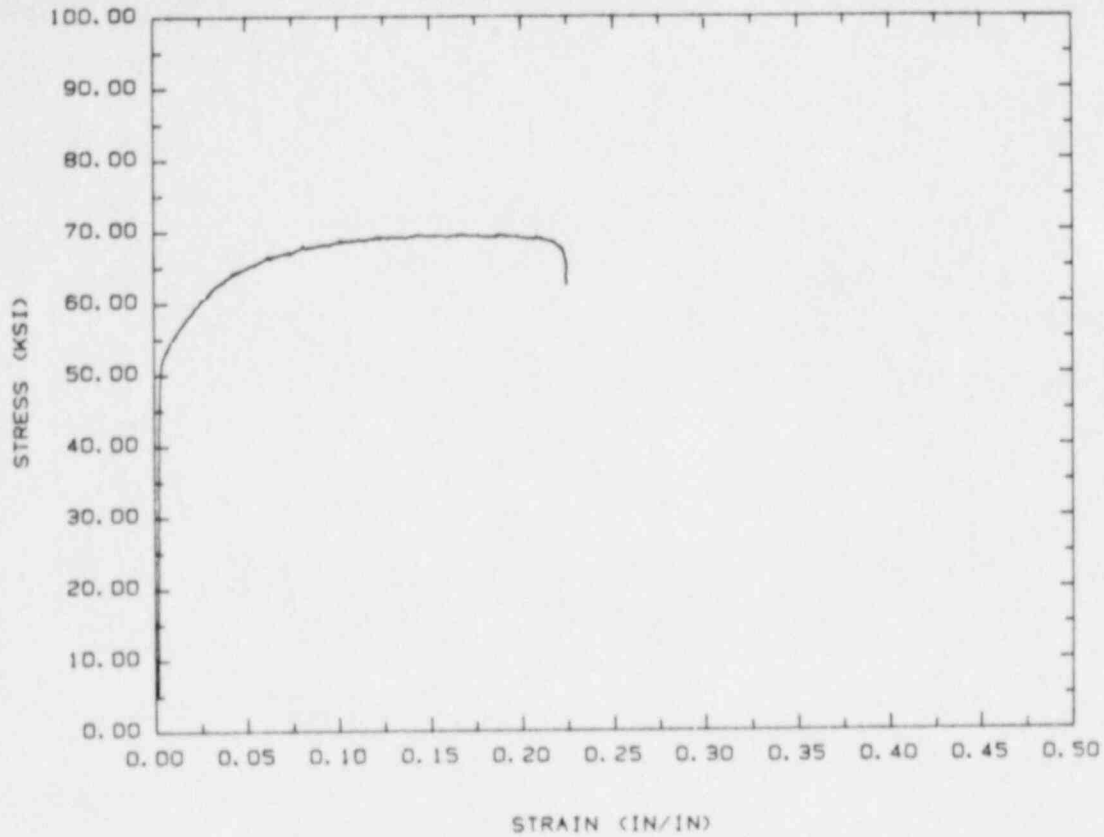


Figure 1 Engineering stress/strain curve for dome material

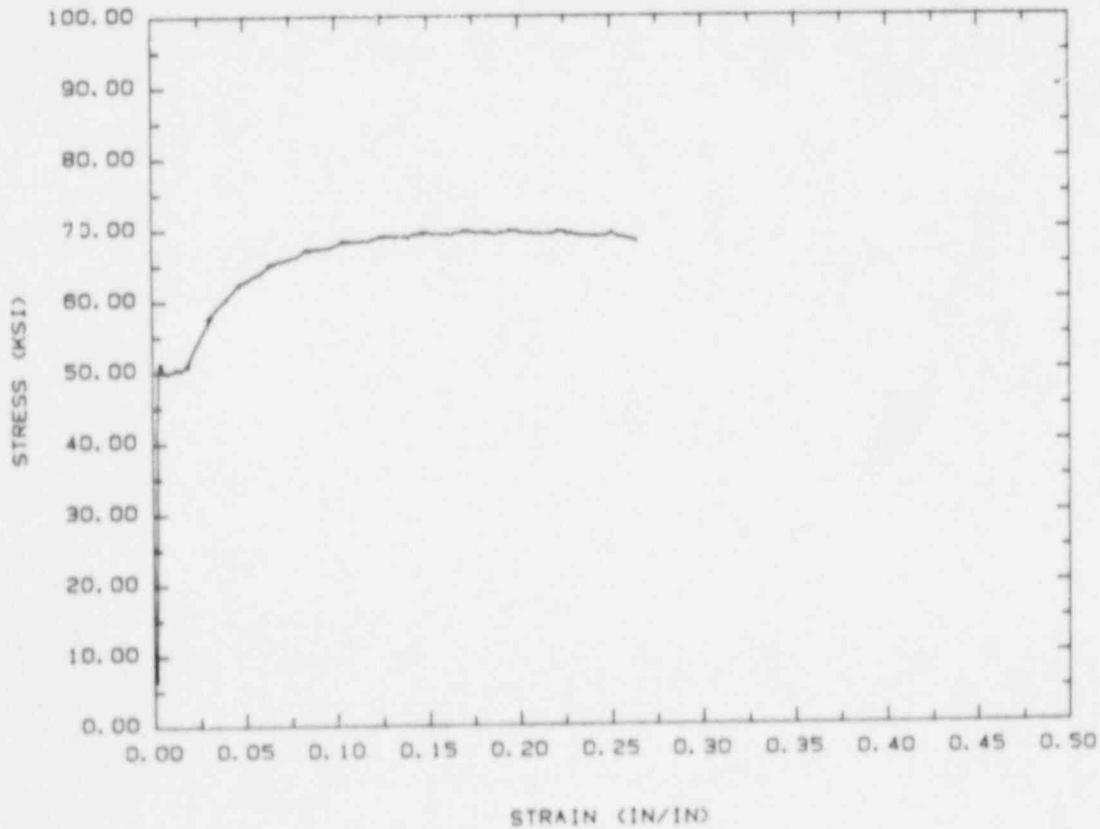


Figure 2 Engineering stress/strain curve for cylinder material

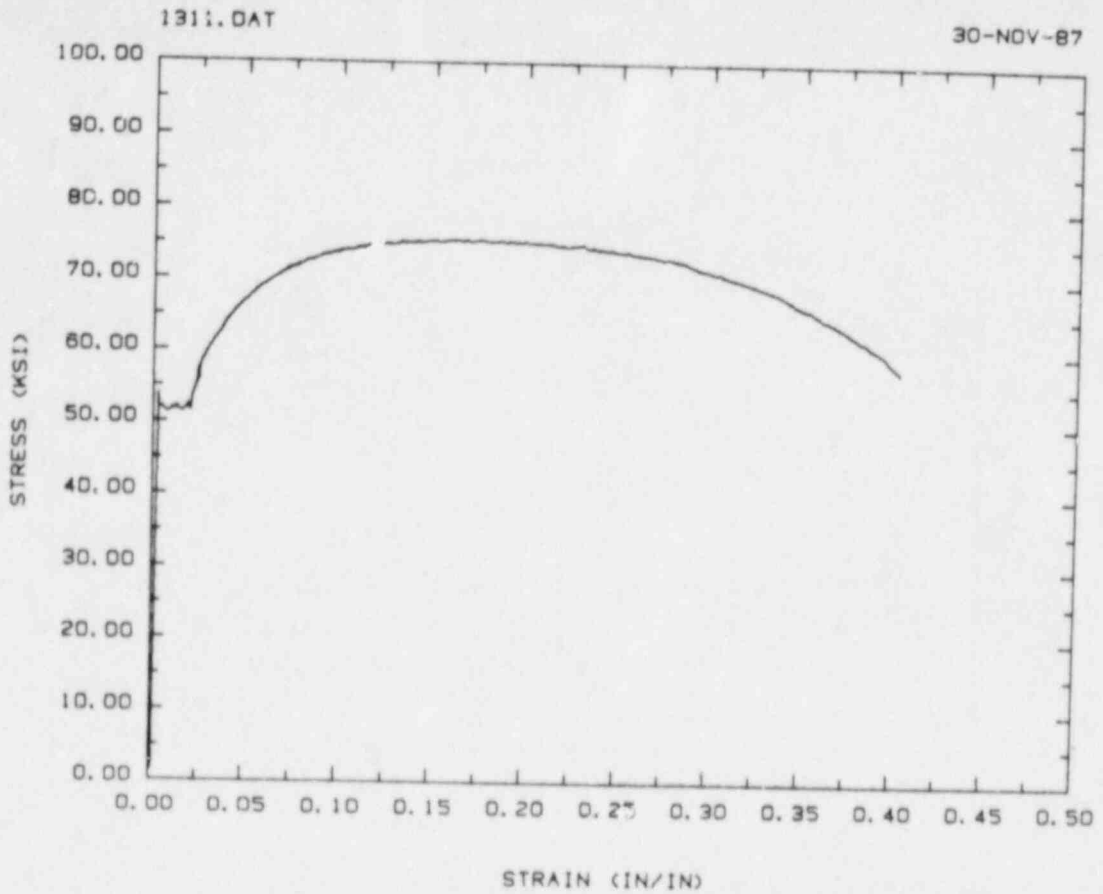


Figure 3 Engineering stress/strain curve for penetration sleeve material

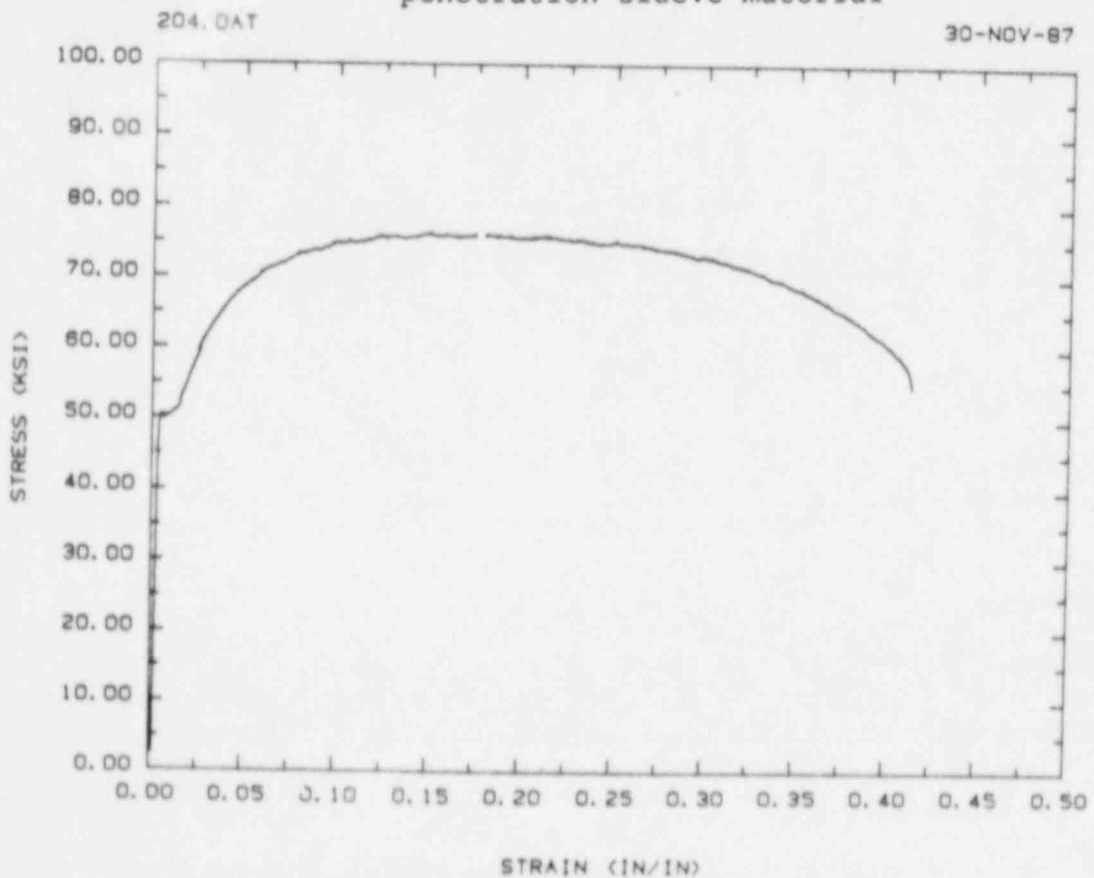


Figure 4 Engineering stress/strain curve for insert plate material

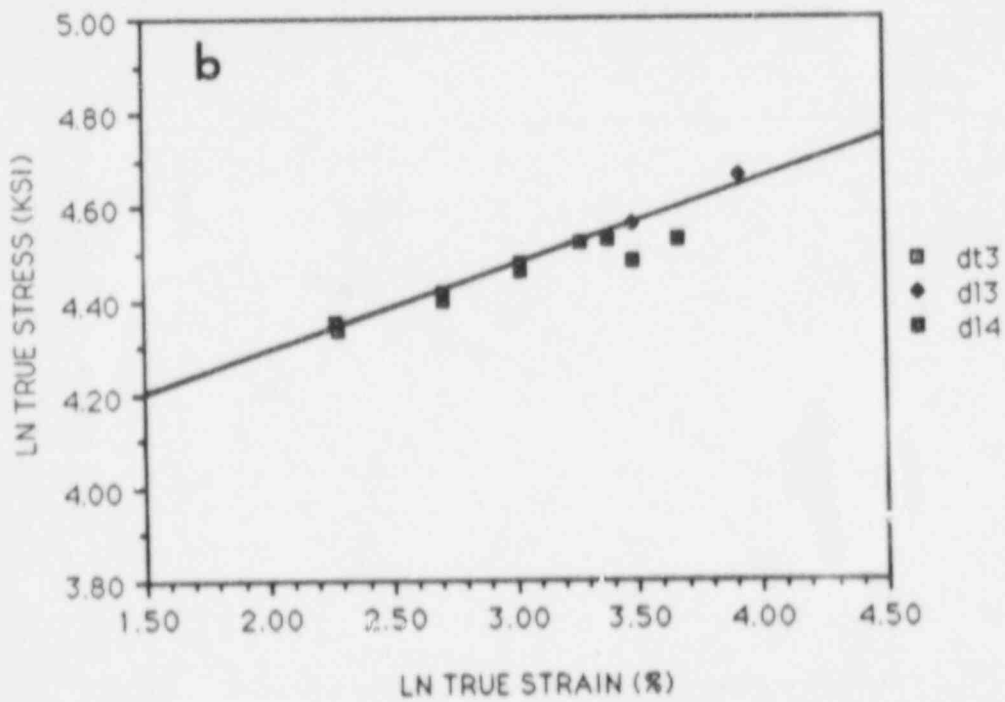
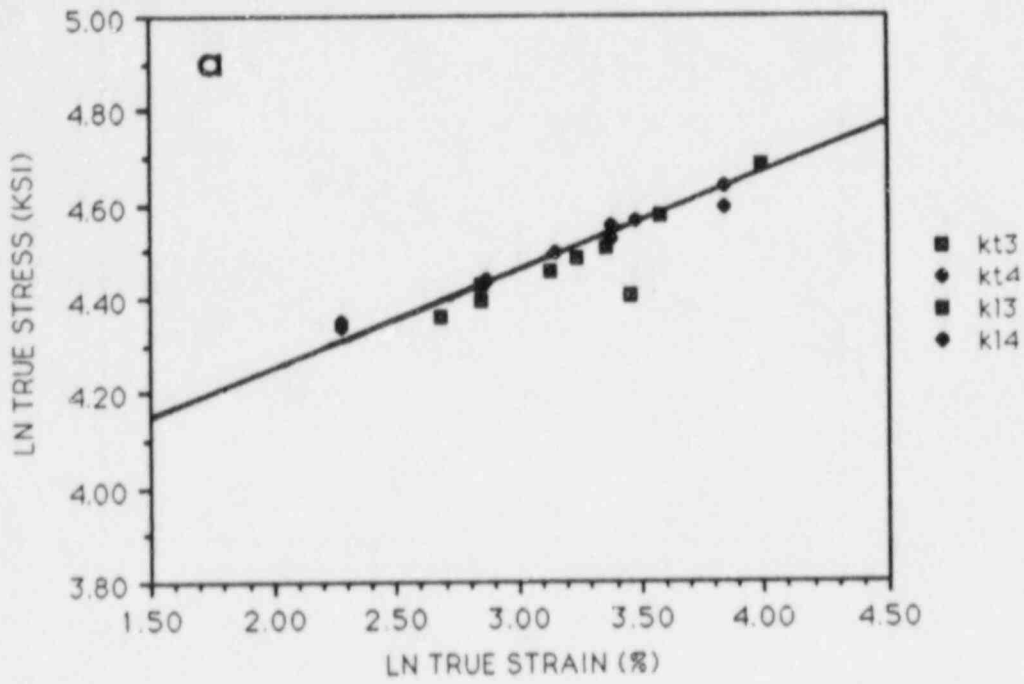


Figure 5 True stress/strain data for dome material a) pre-dished condition, b) post-dished condition

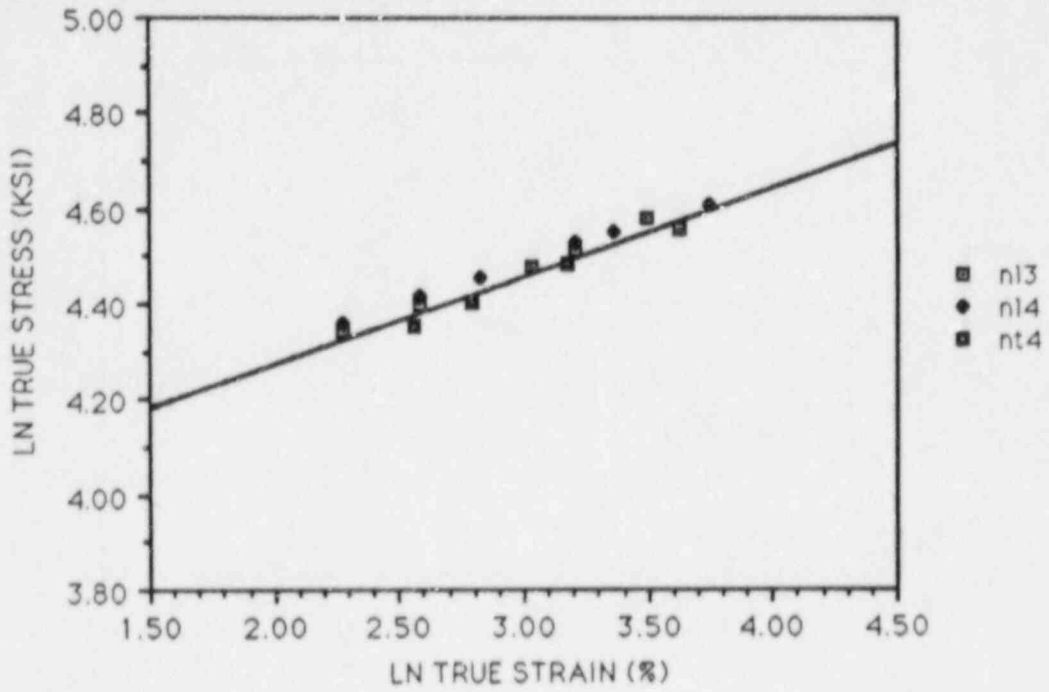


Figure 6 True stress/strain data for cylinder material

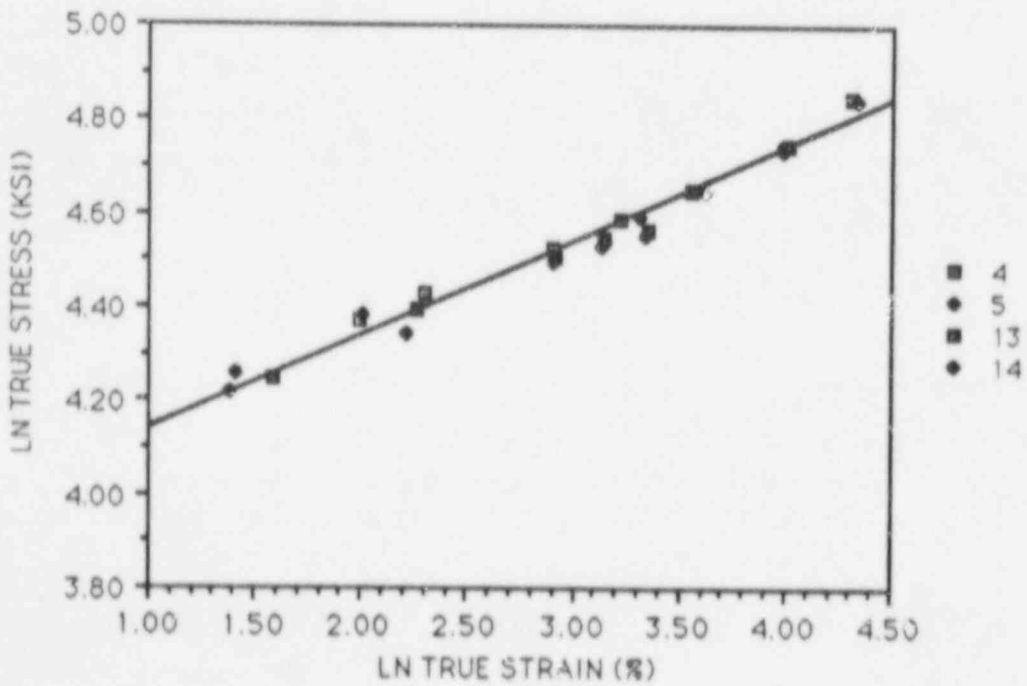


Figure 7 True stress/strain data for penetration sleeve material

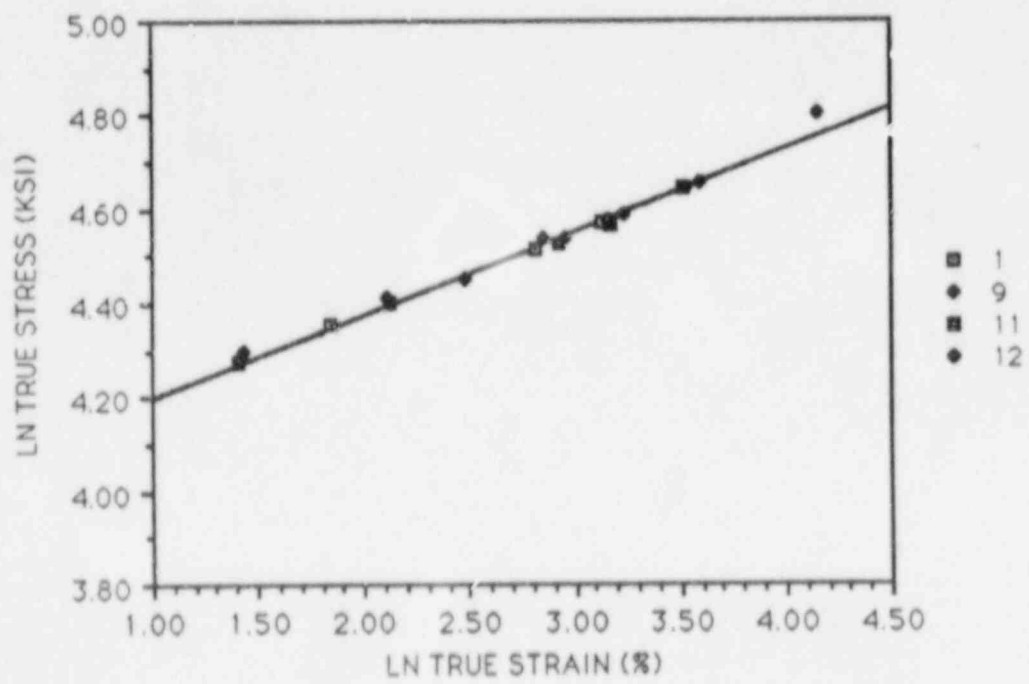
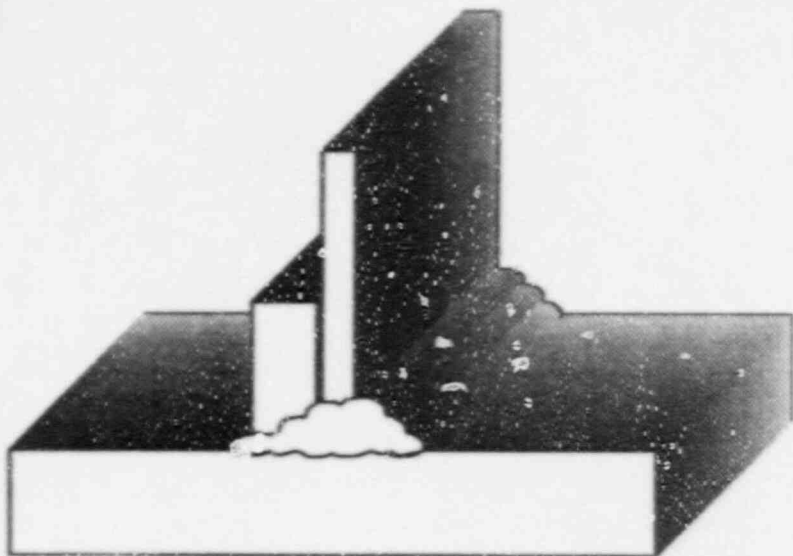
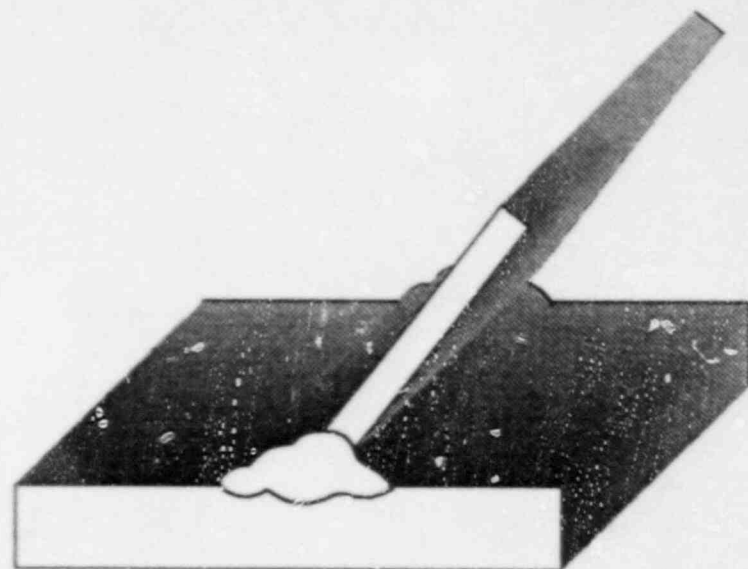


Figure 8 True stress/strain data for insert plate material

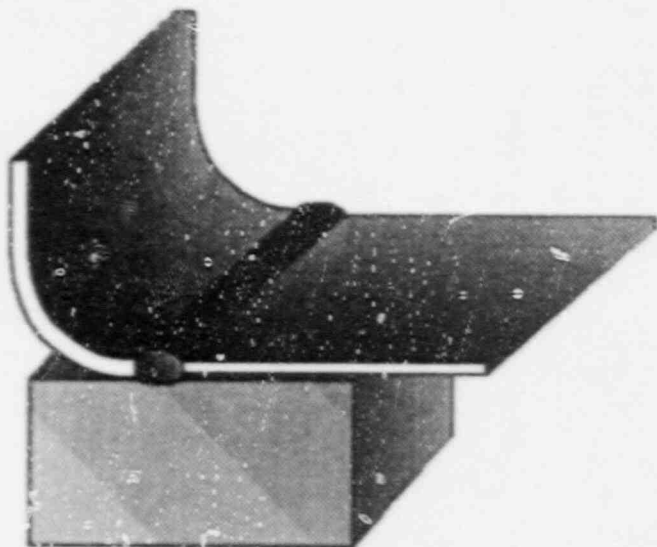


a) Penetration sleeve (1.3" reduced to 0.5") to insert plate (0.2"), with backup bar.

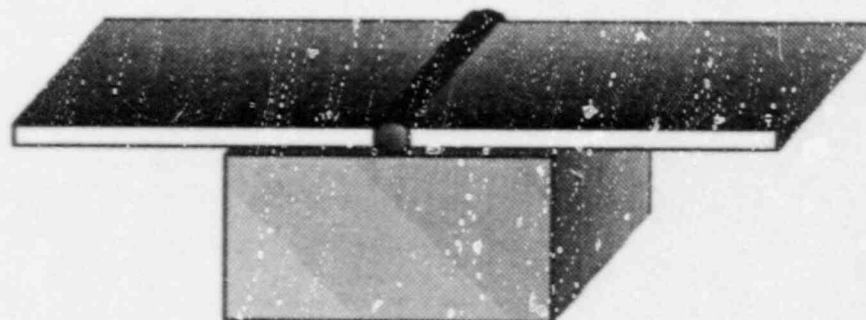


b) Equipment hatch dome (0.2") to tension ring.

20

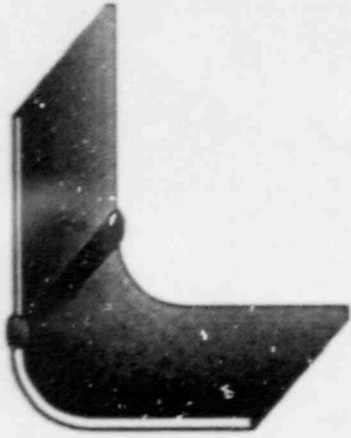


c) Bottom corner to 0.07" bottom with 1" x 2" backing bar (does not illustrate 0.07" cylinder to bottom corner weld).

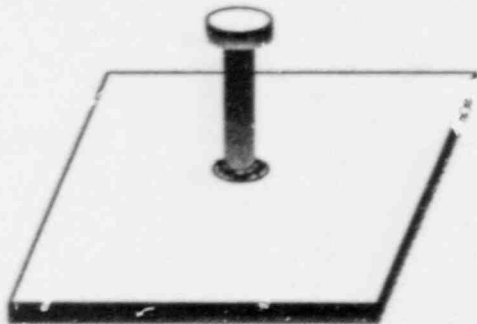


d) 0.07" bottom to 0.07" bottom with 1" x 2" backing bar.

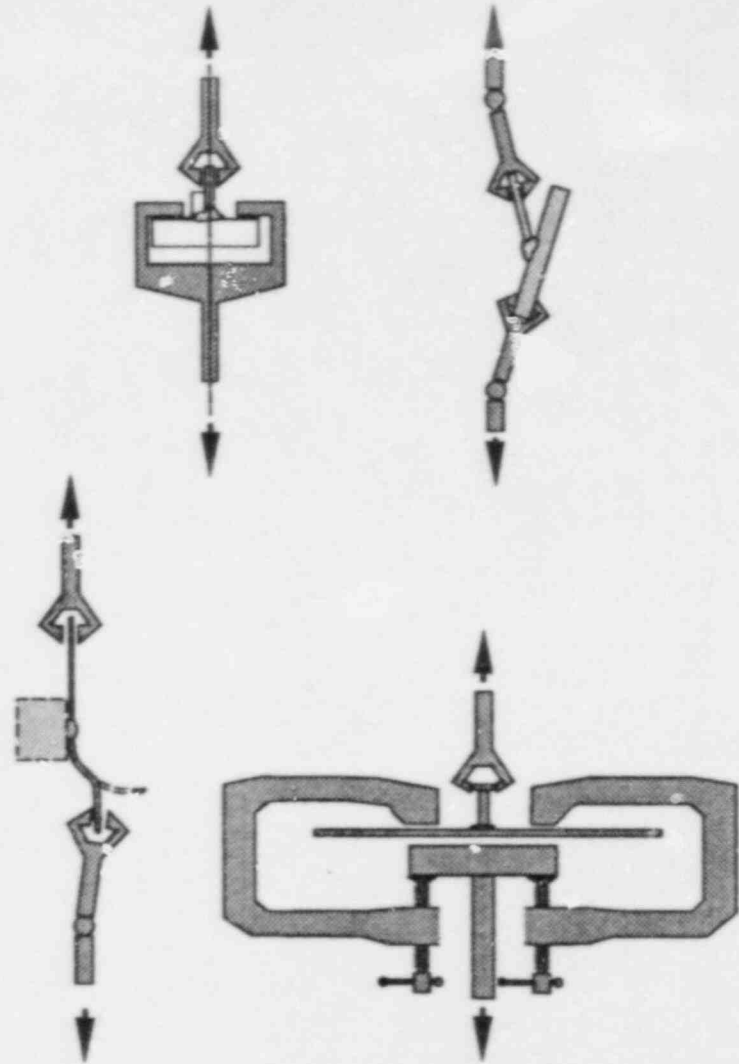
Figure 9 Additional weld specimen geometries.



e) 0.07" cylinder to bottom corner.



f) Scud welds, 1/8" shaft to plates 0.07" thick, 0.09" thick and 0.2" thick.



g) Details of loading geometries.

Figure 9 Additional weld specimen geometries.

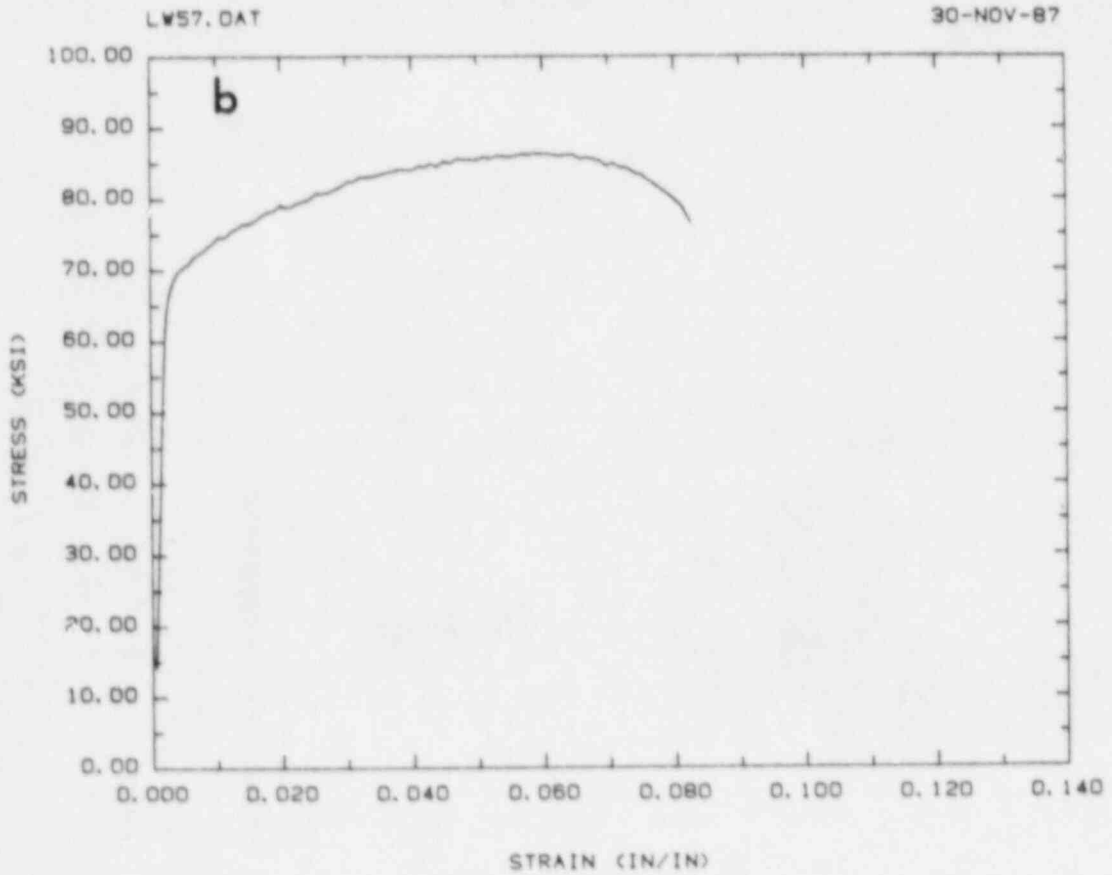
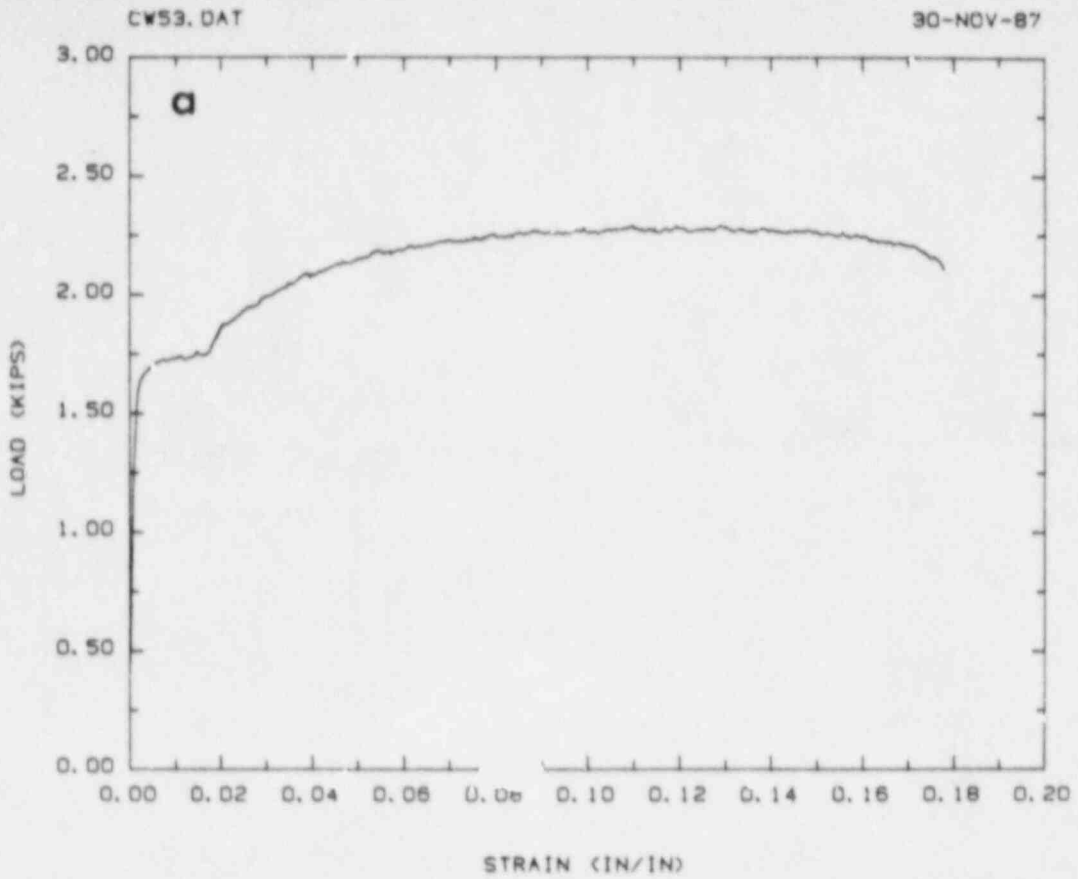


Figure 10 a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/cylinder specimen



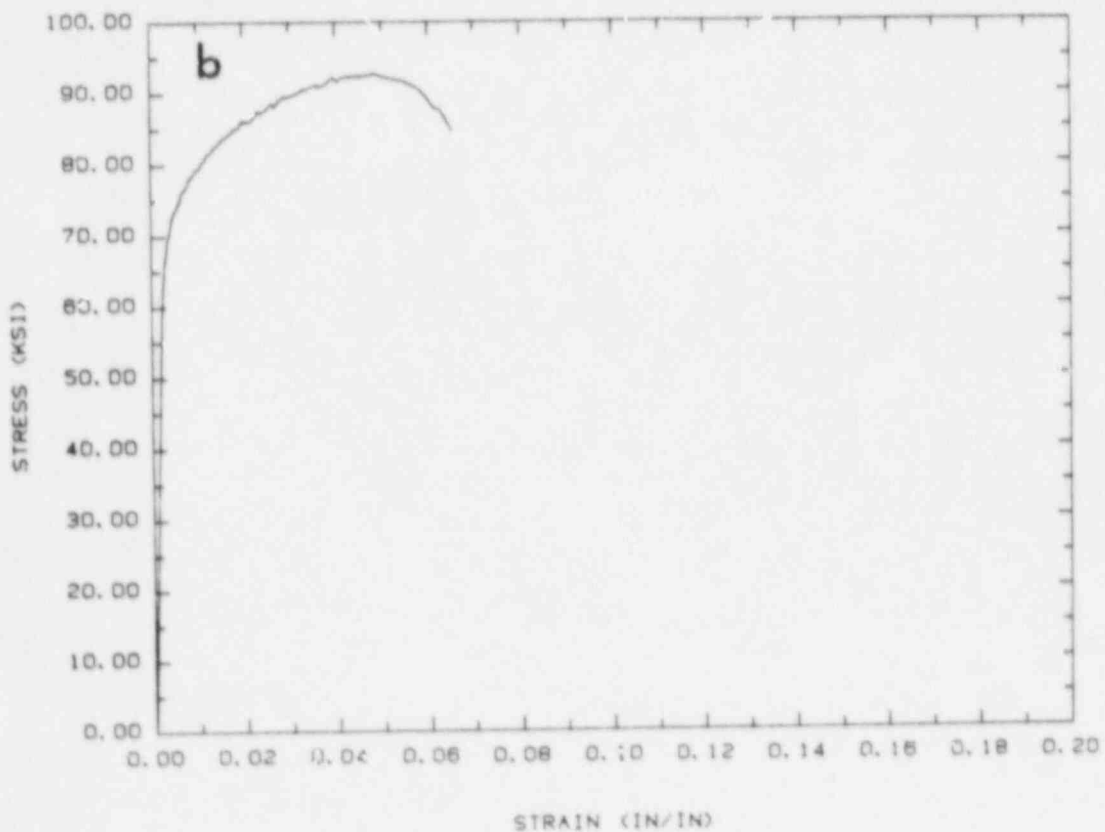
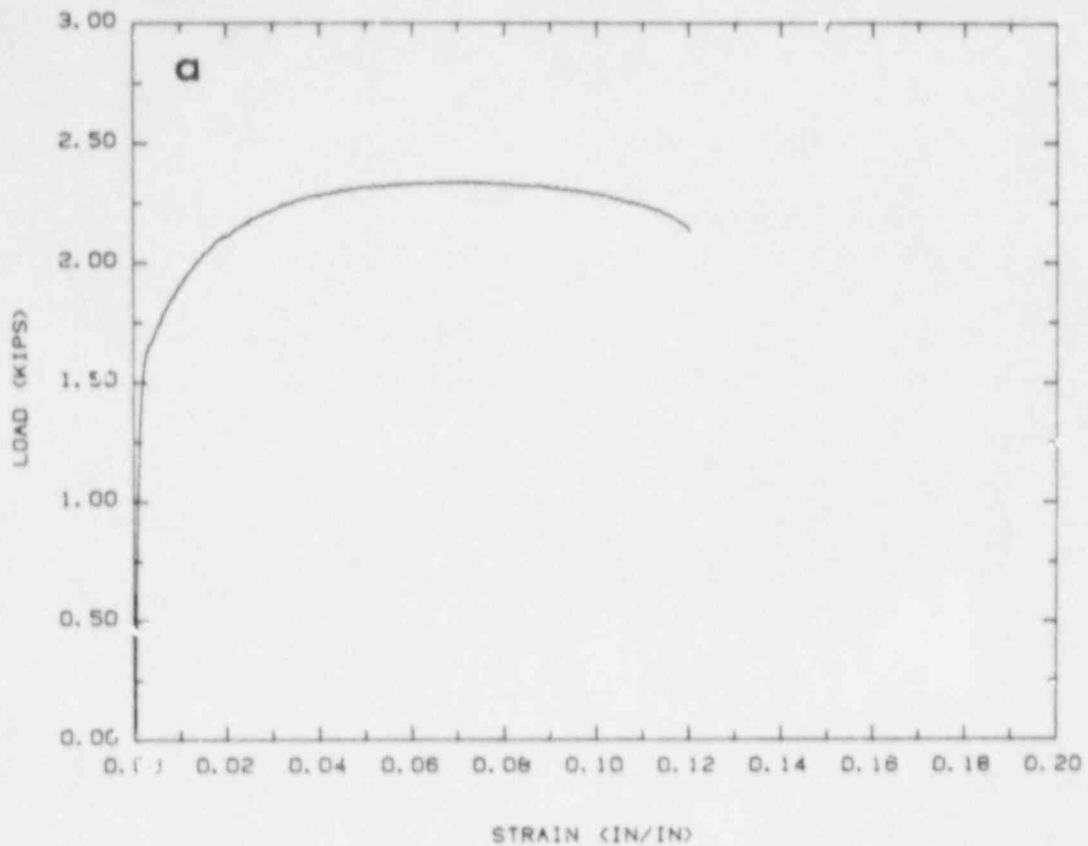


Figure 11 a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/insert plate specimen

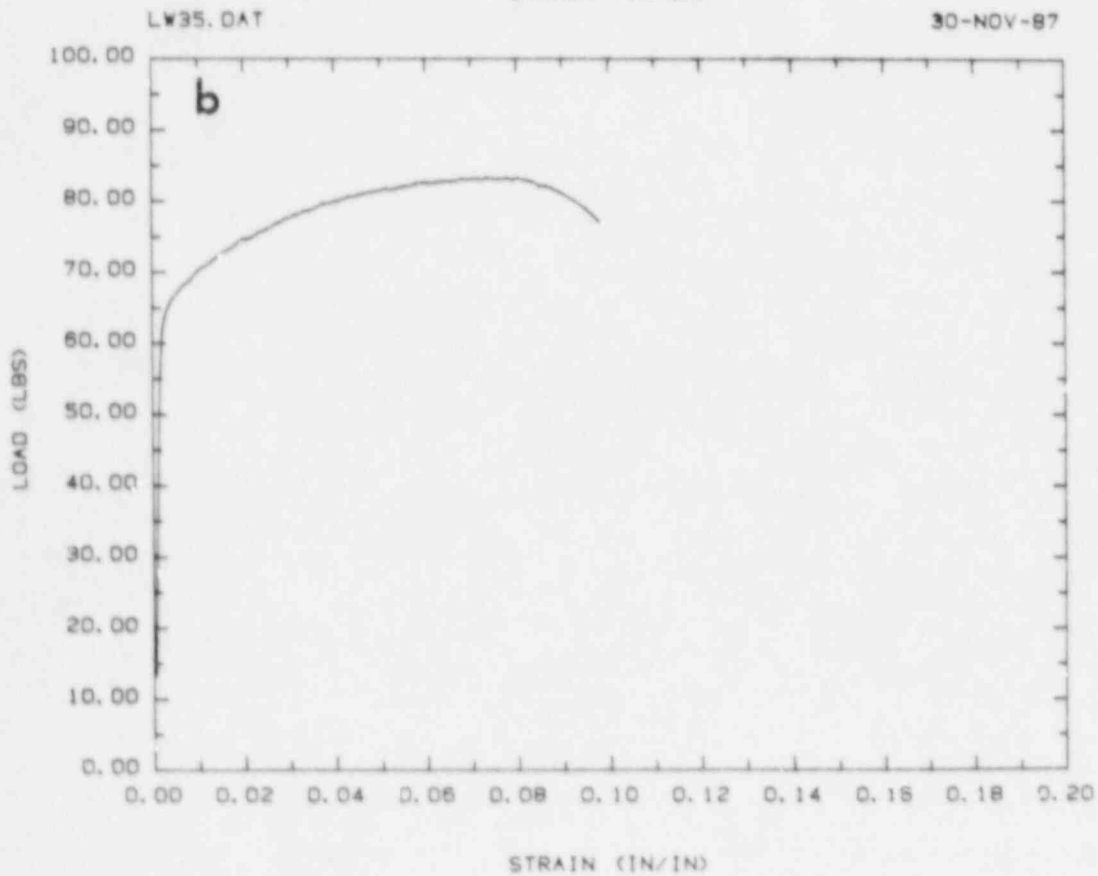
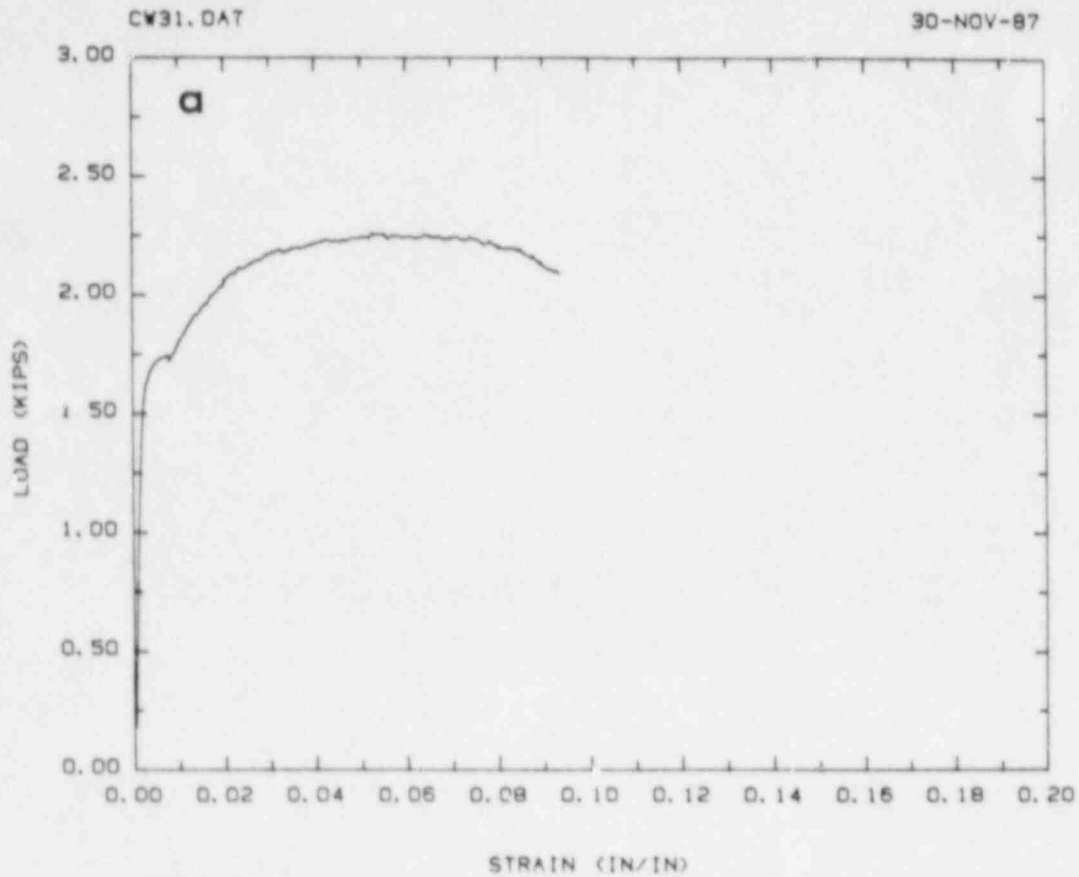


Figure 12 a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for cylinder/dome specimen

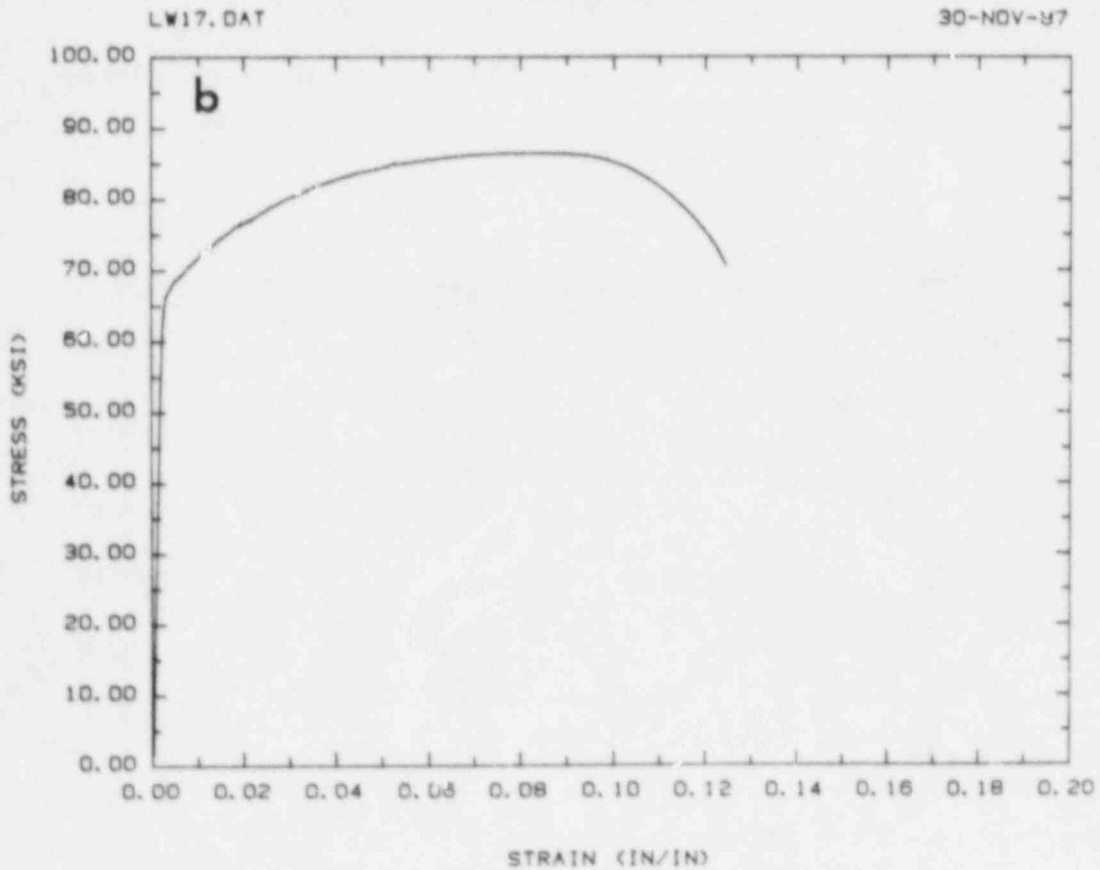
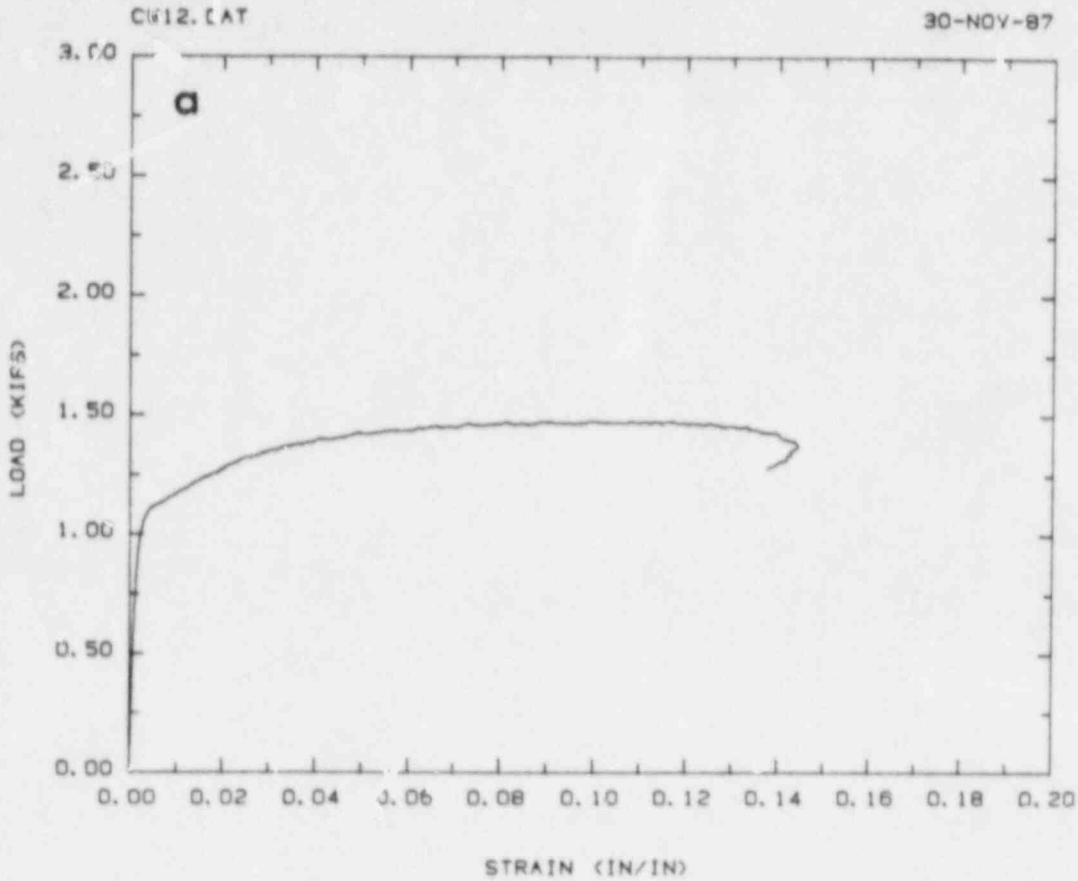


Figure 13 a) Cross-weld load/engineering strain curve and b) all weld metal engineering stress/strain curve for dome/dome specimen

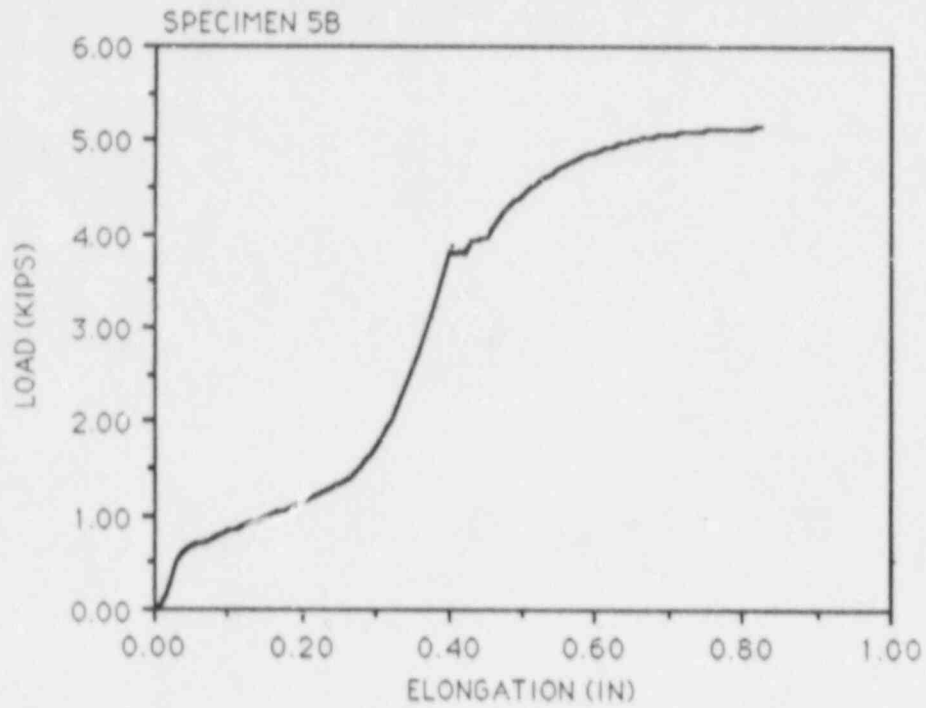


Figure 14 Cross-weld load/elongation curve for cylinder/bottom corner specimen

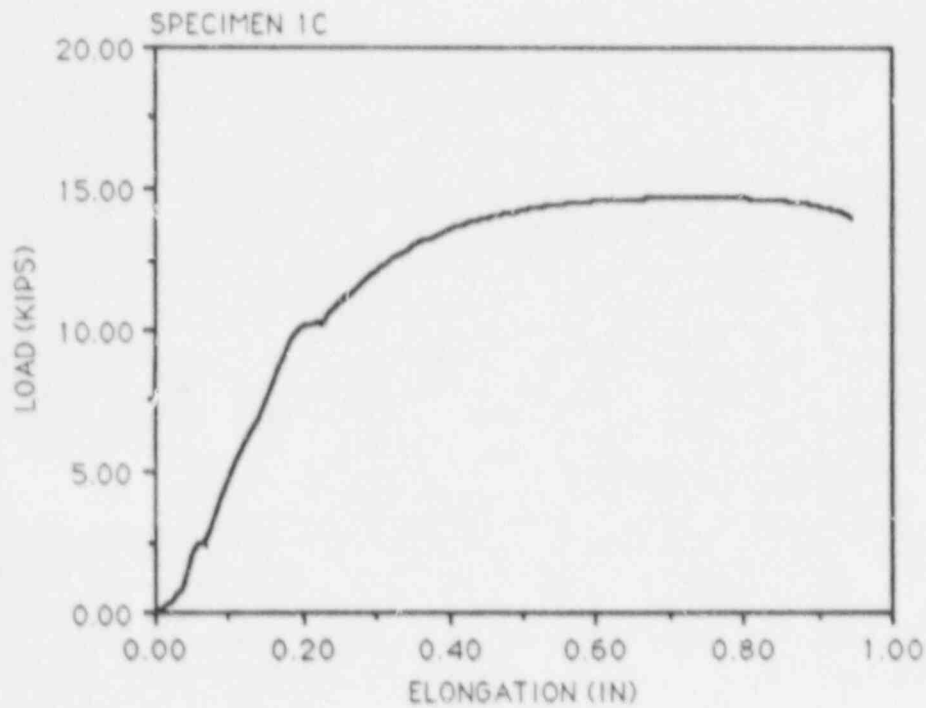


Figure 15 Cross-weld load/elongation curve for penetration sleeve/insert plate specimen

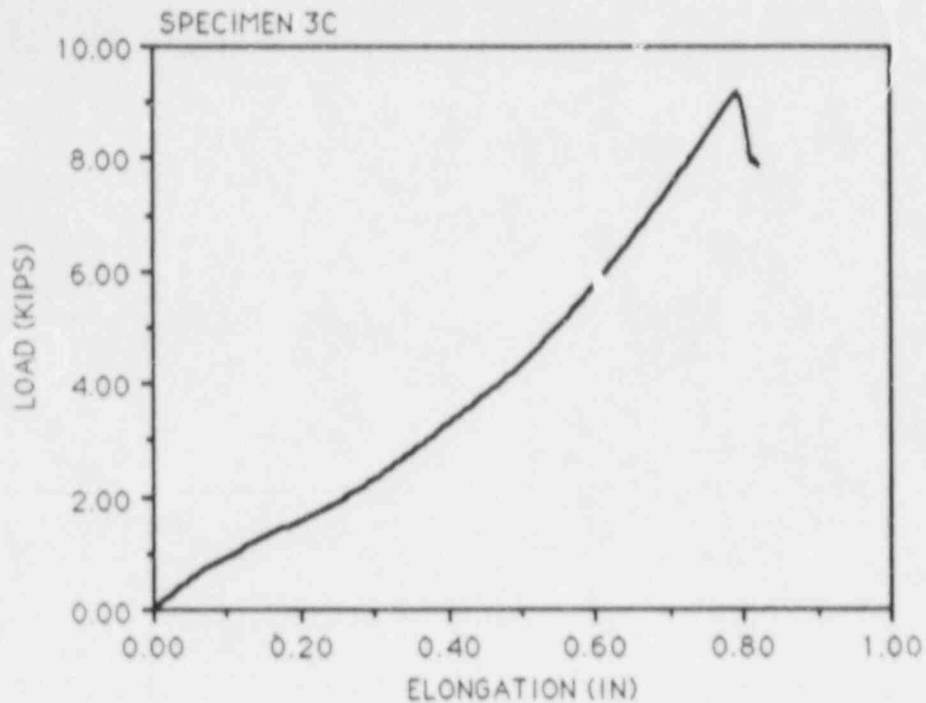


Figure 16 Cross-weld load/elongation curve for equipment hatch dome to tension ring specimen

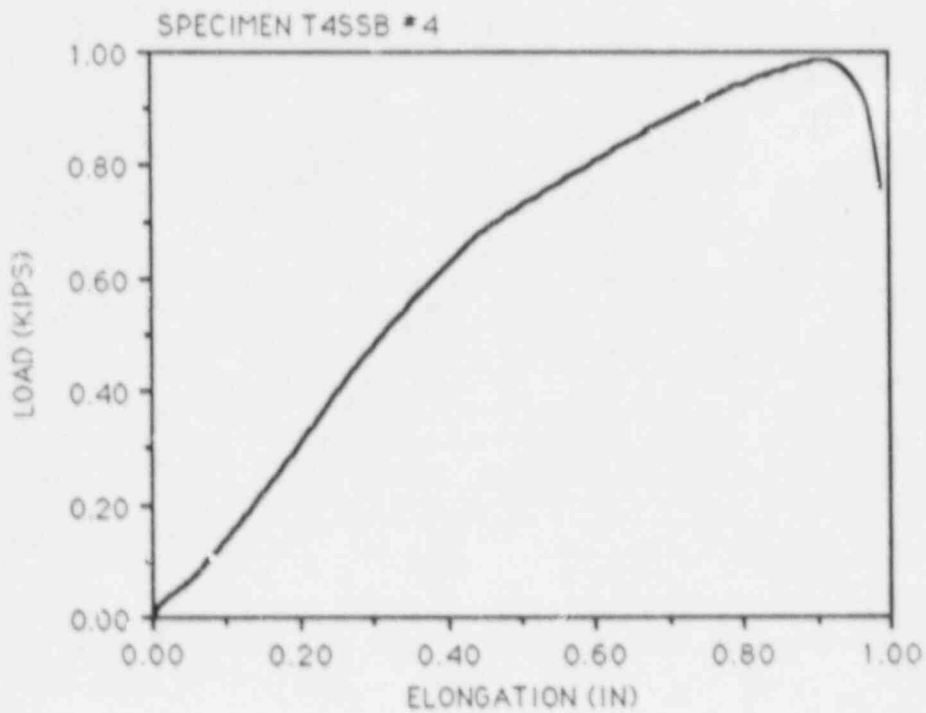


Figure 17 Load/elongation curve for stud to cylinder weld

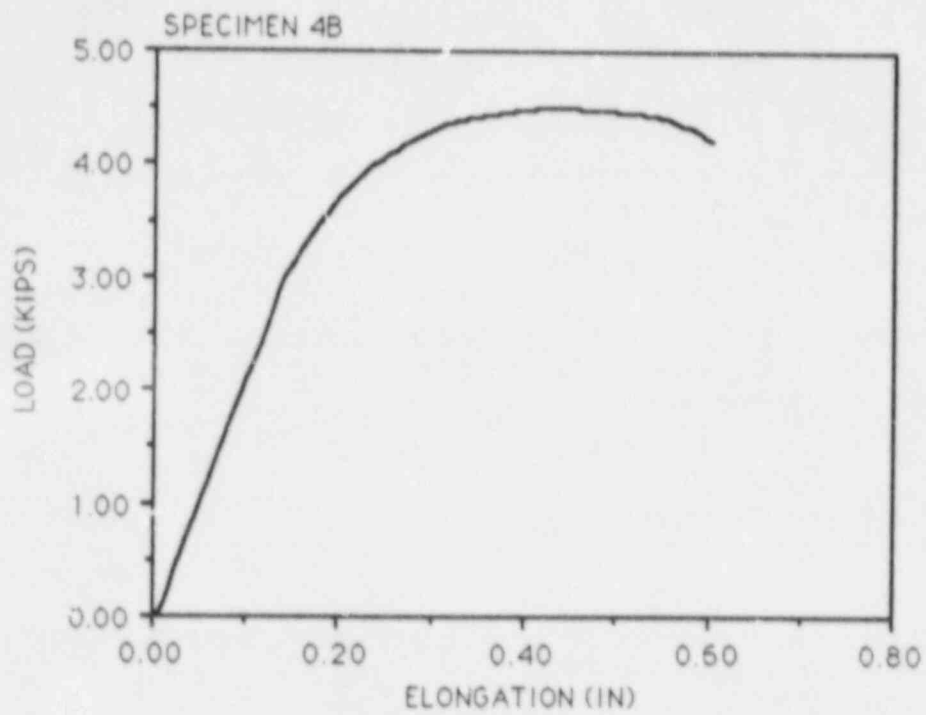


Figure 18 Cross-weld load/elongation curve for bottom corner to bottom with backing bar

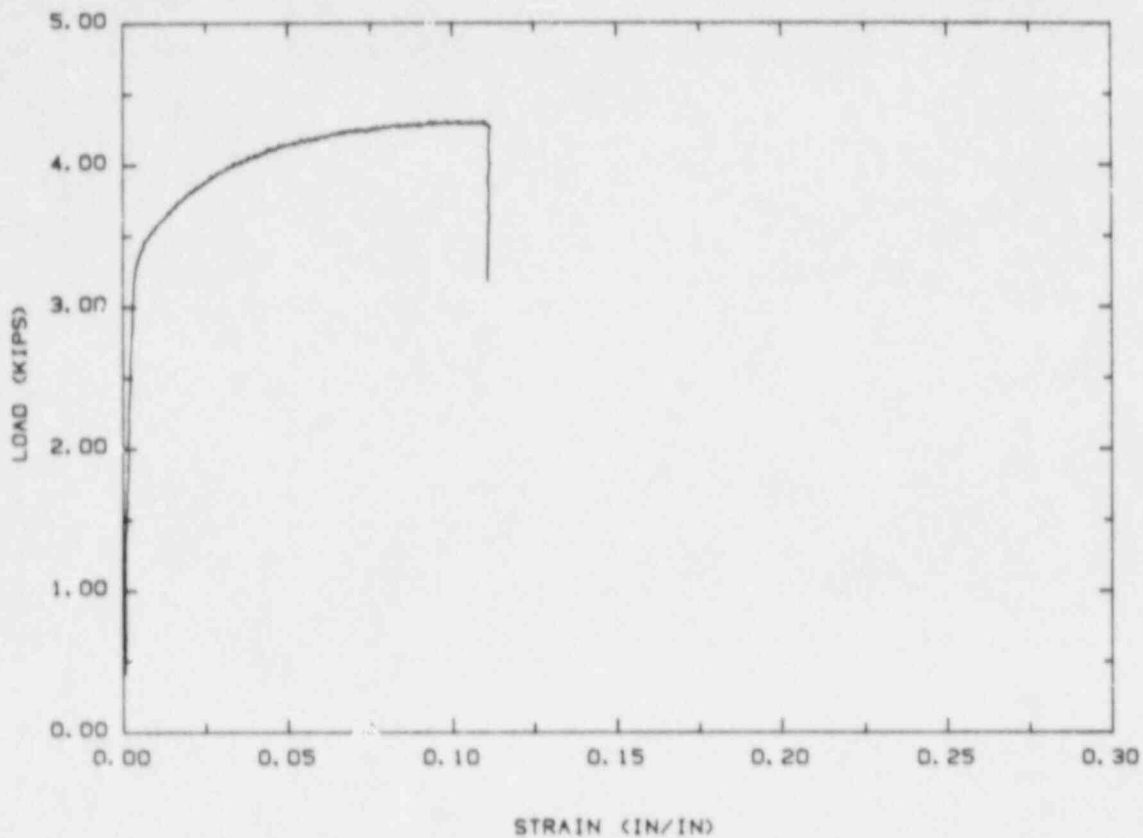


Figure 19 6 mm rebar load/engineering strain curve

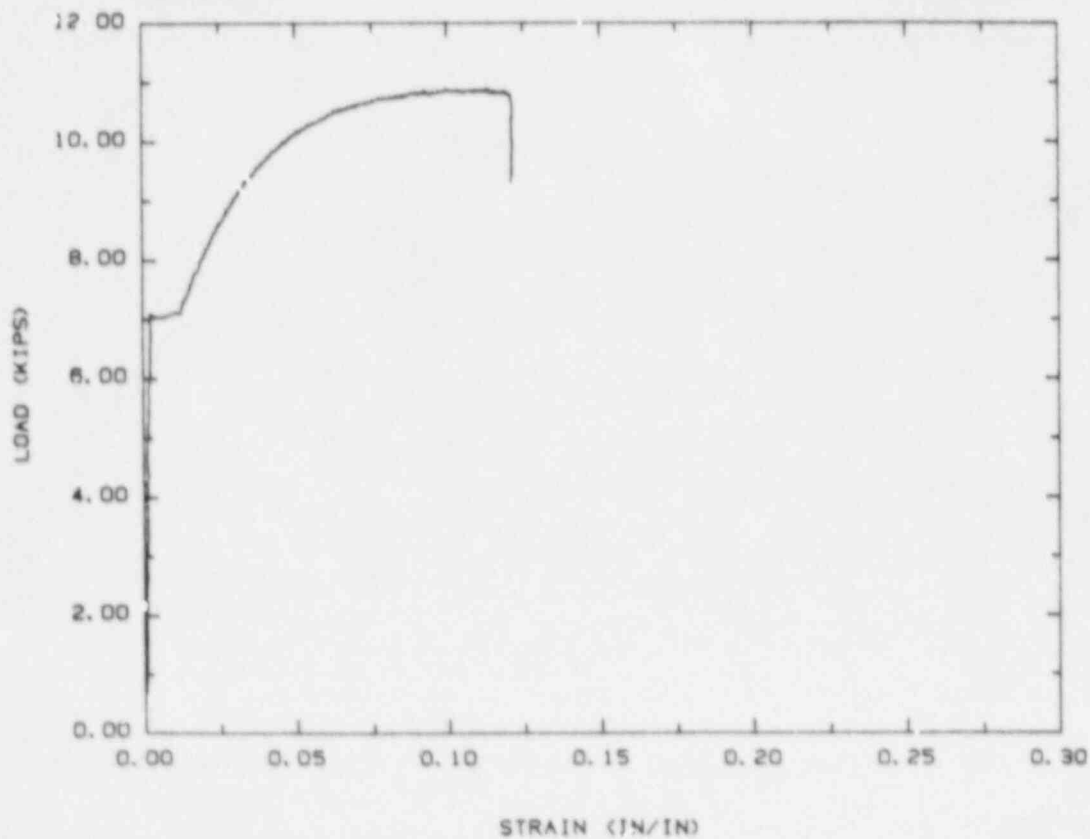


Figure 20 #3 rebar load/engineering strain curve

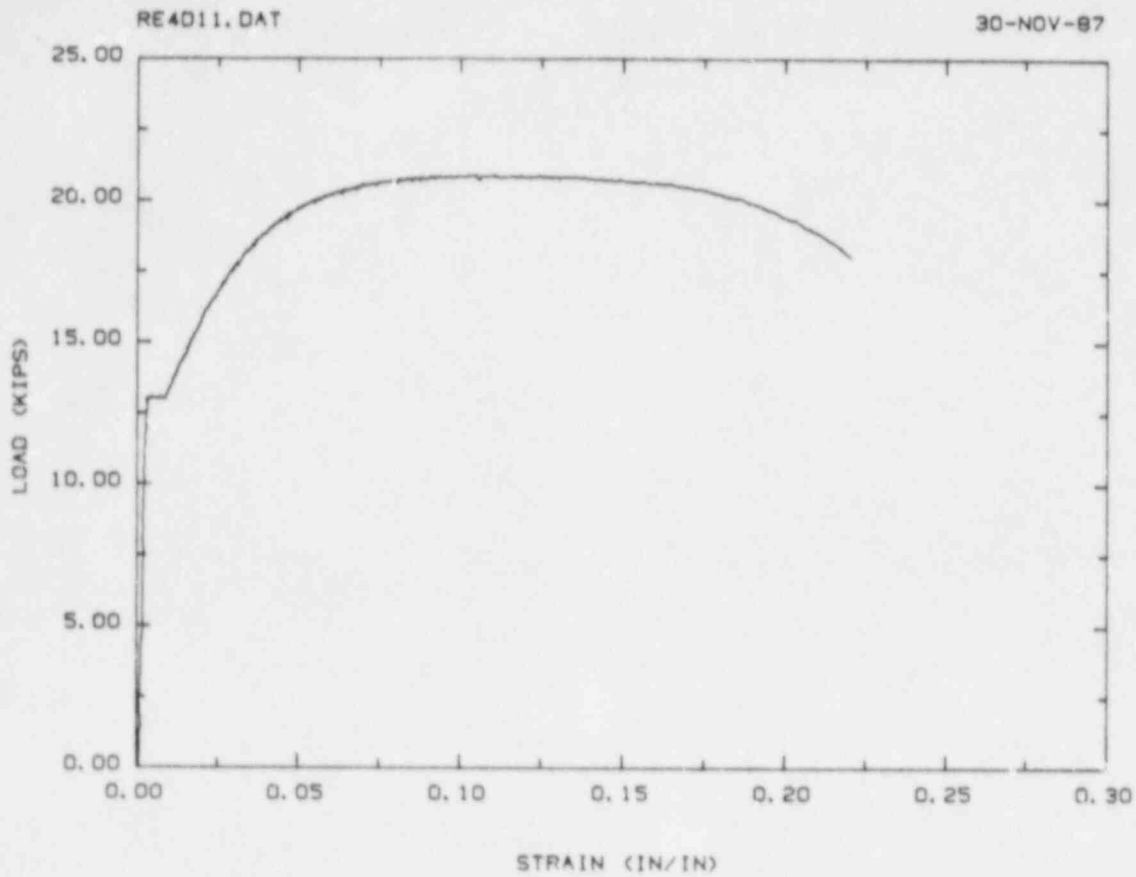


Figure 21 #4 rebar load/engineering strain curve

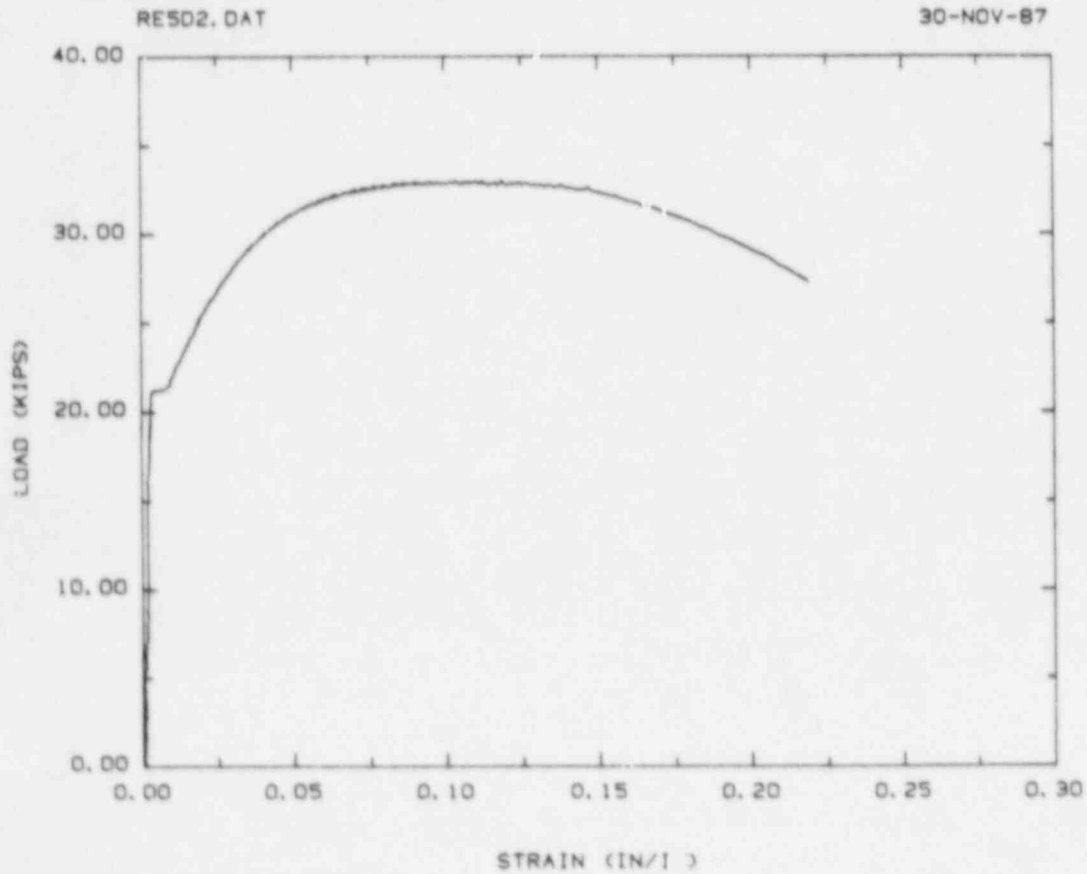


Figure 22 #5 rebar load/engineering strain curve



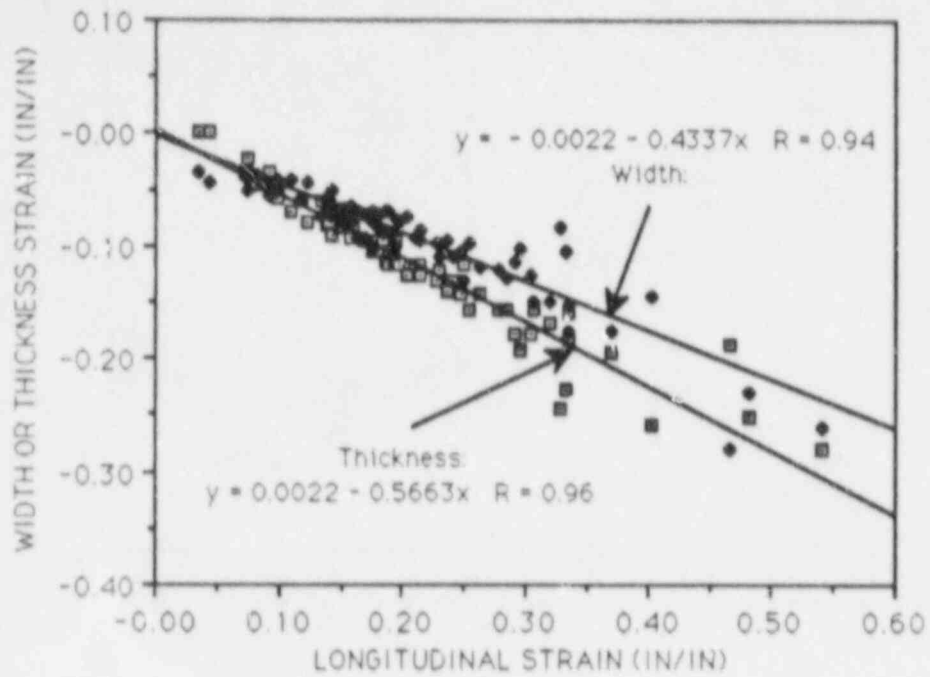


Figure 23 True width and thickness strain vs.true longitudinal strain for dome material

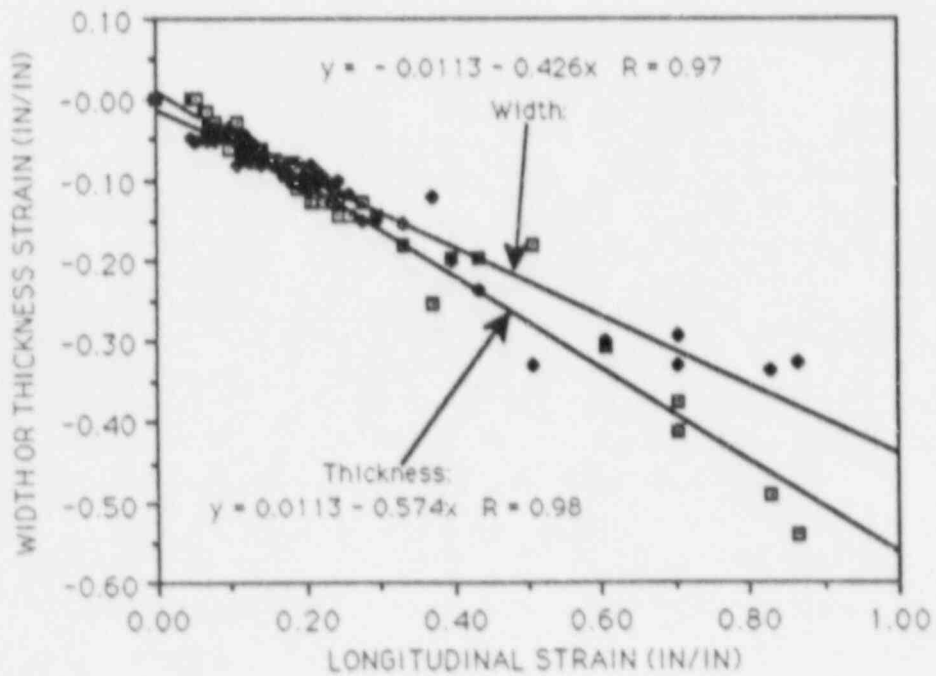


Figure 24 True width and thickness strain vs. true longitudinal strain for cylinder material

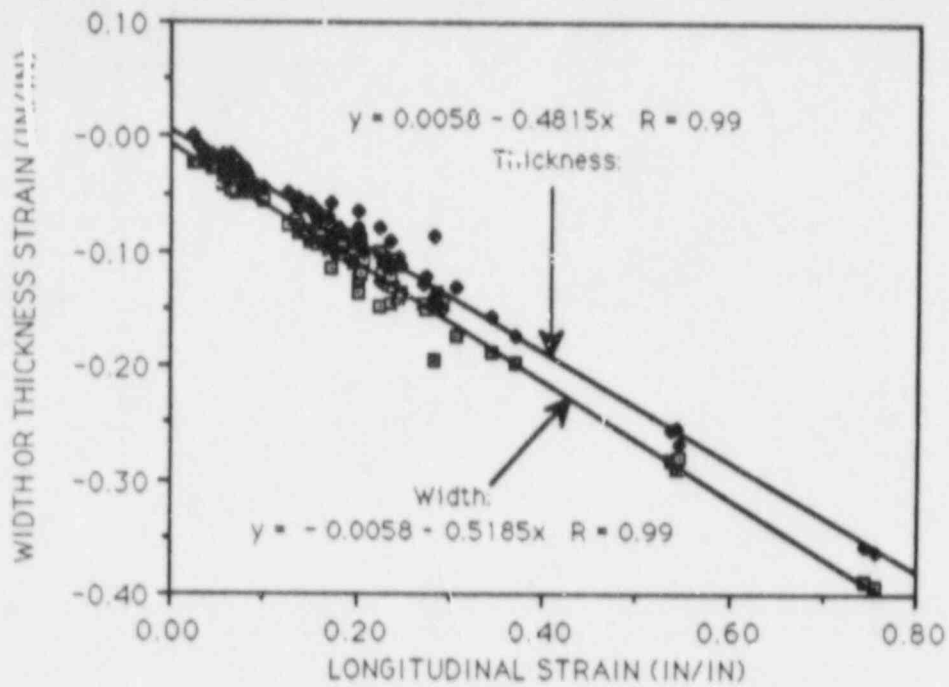


Figure 25 True width and thickness strain vs. true longitudinal strain for penetration sleeve material

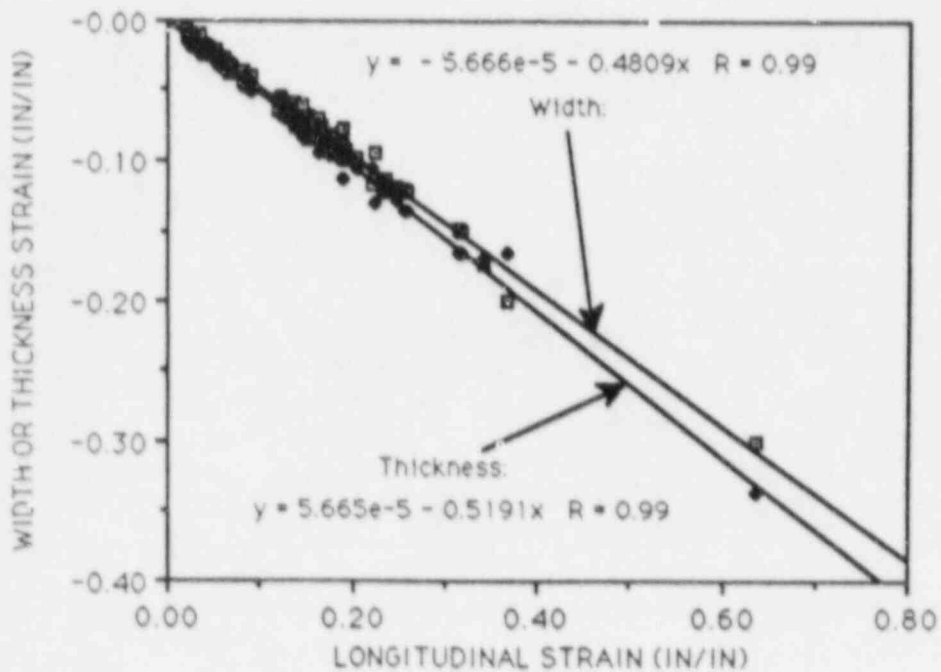


Figure 26 True width and thickness strain vs. true longitudinal strain for insert plate material

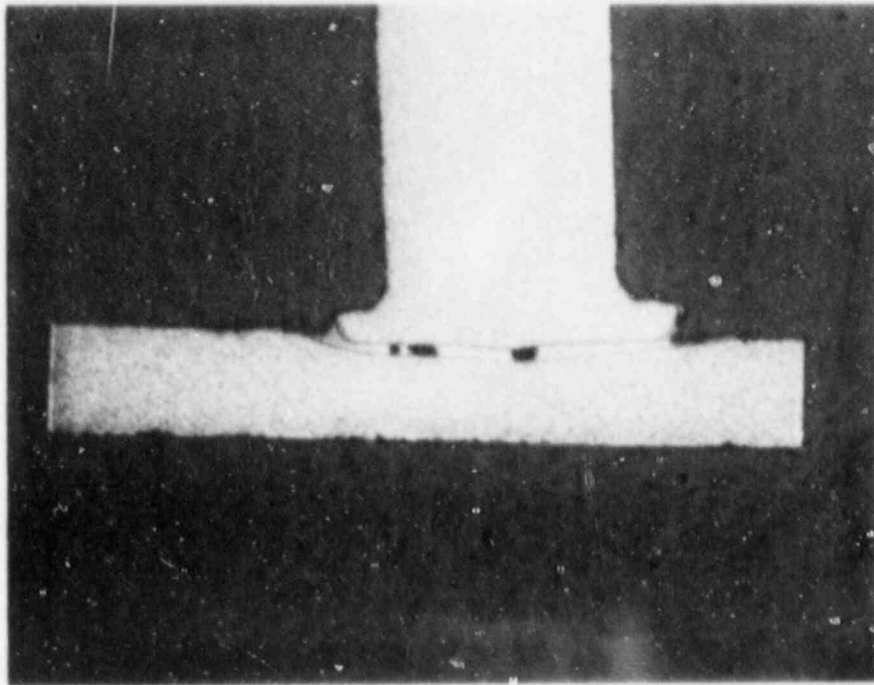


Figure 27 Cross-section of stud weld to cylinder material (8.5x)

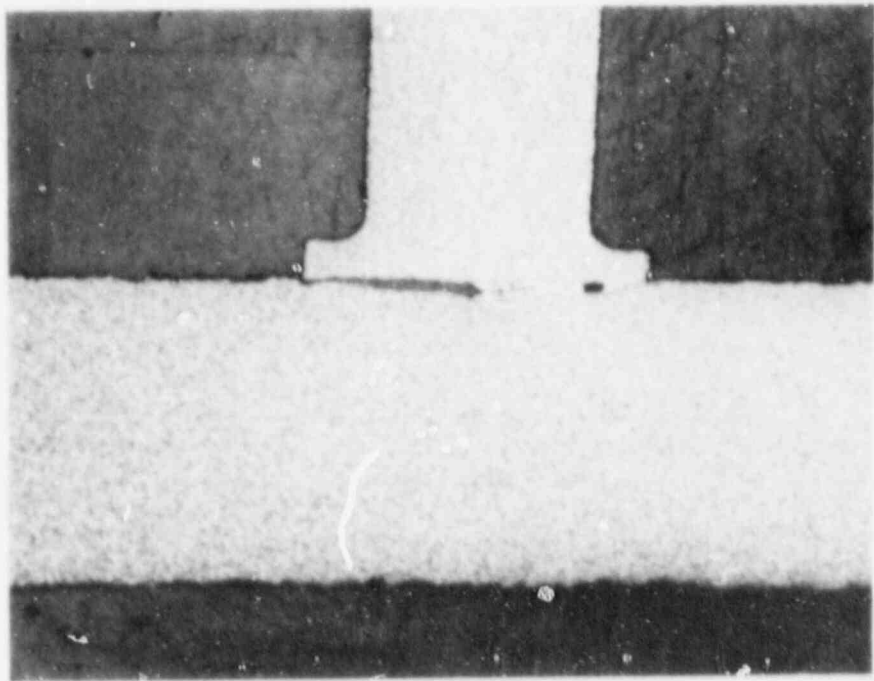


Figure 28 Cross-section of stud weld to insert plate material (8.5x)

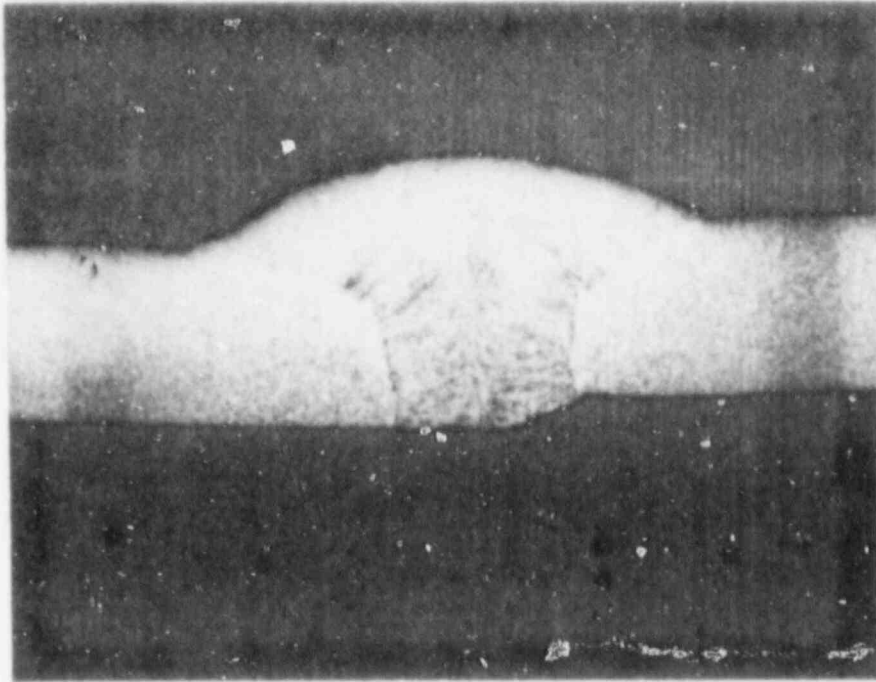


Figure 29 Cross-section of weld between dome sections (flat position 10x)

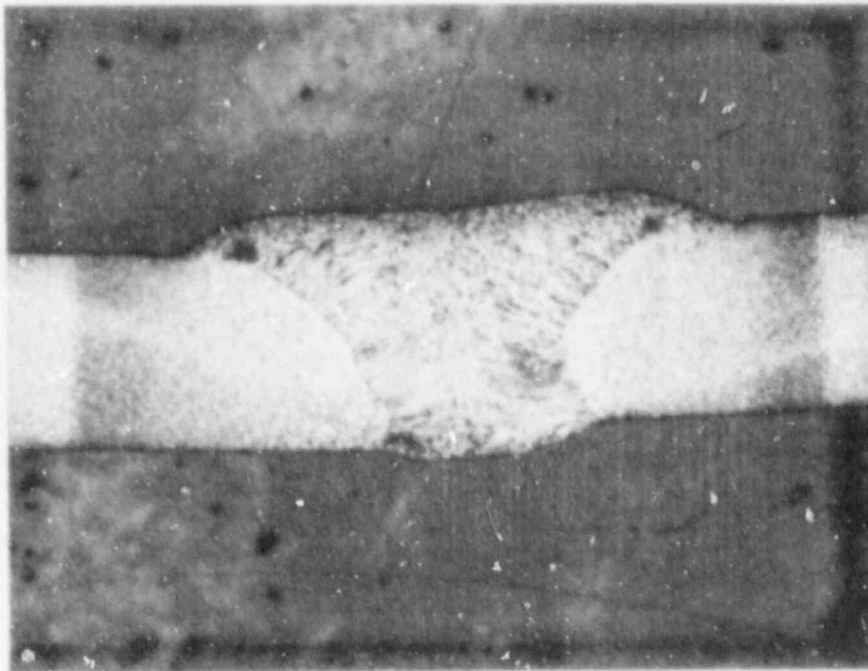


Figure 30 Cross-section of weld between dome sections (vertical position 12x)

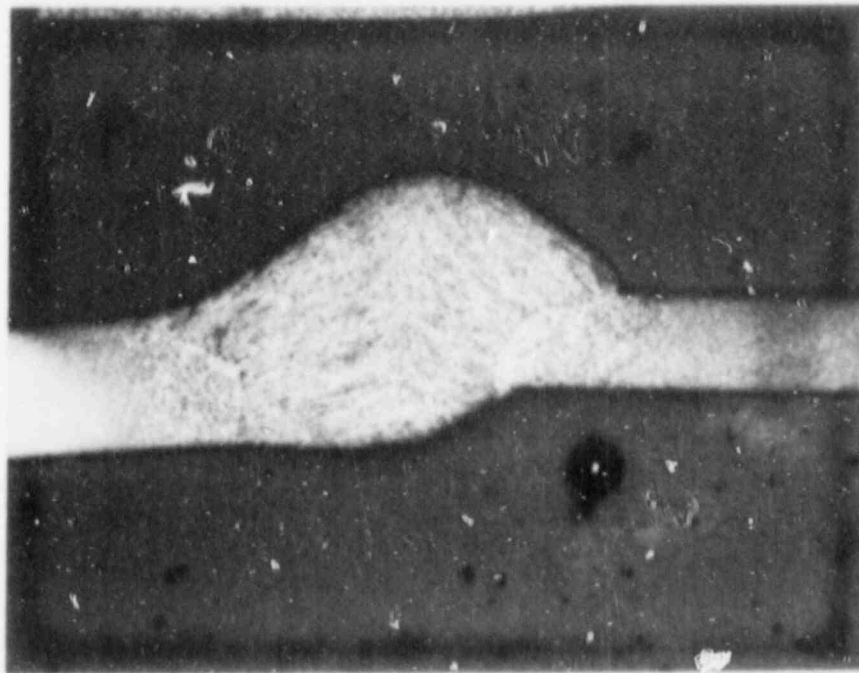


Figure 31 Cross-section of weld between dome and cylinder (horizontal position 8.5x)

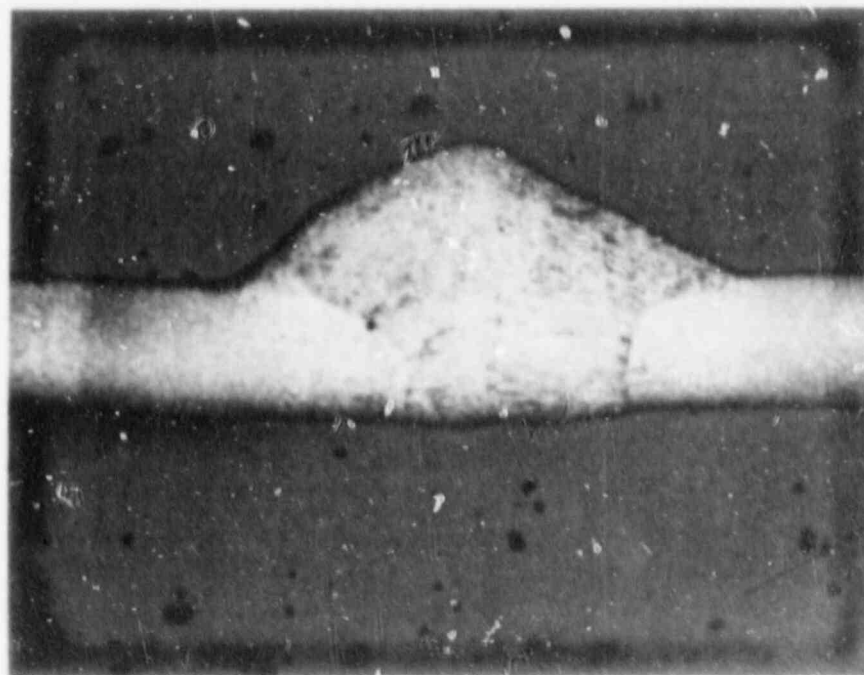


Figure 32 Cross-section of weld between cylinder sections (horizontal position 10x)

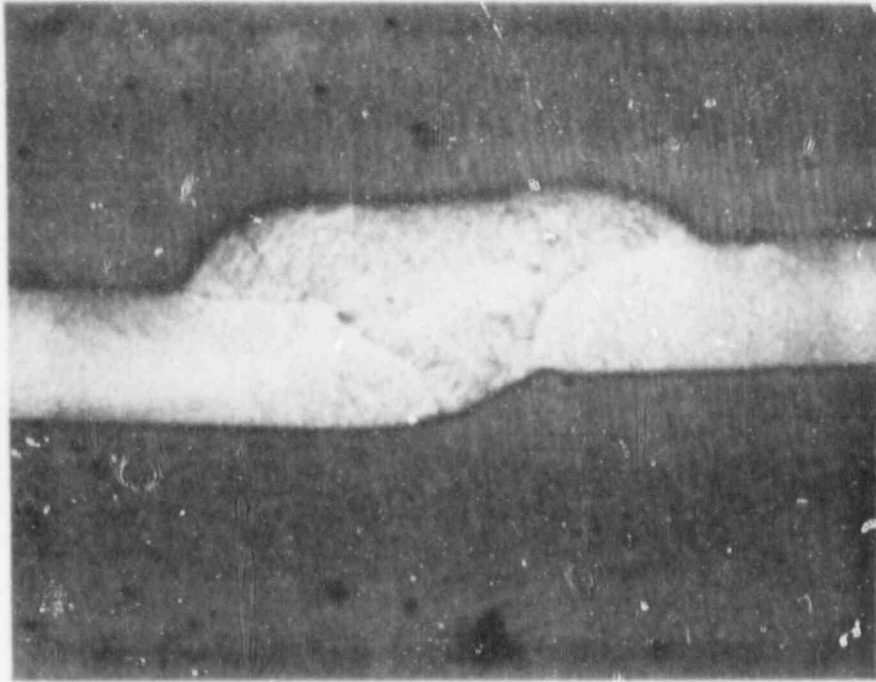


Figure 33 Cross-section of weld between cylinder sections (vertical position 10x)

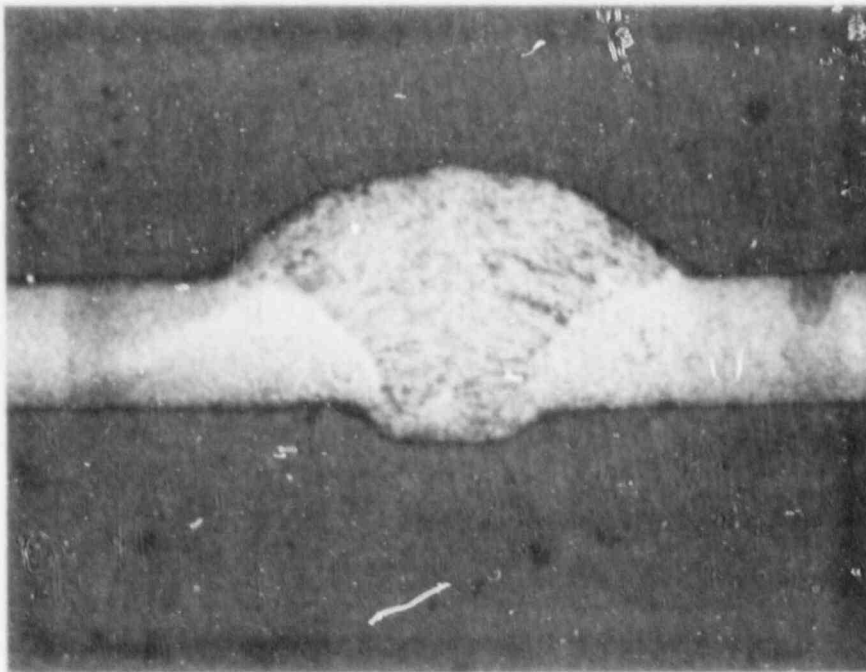


Figure 34 Cross-section of weld between cylinder sections (flat position 10x)

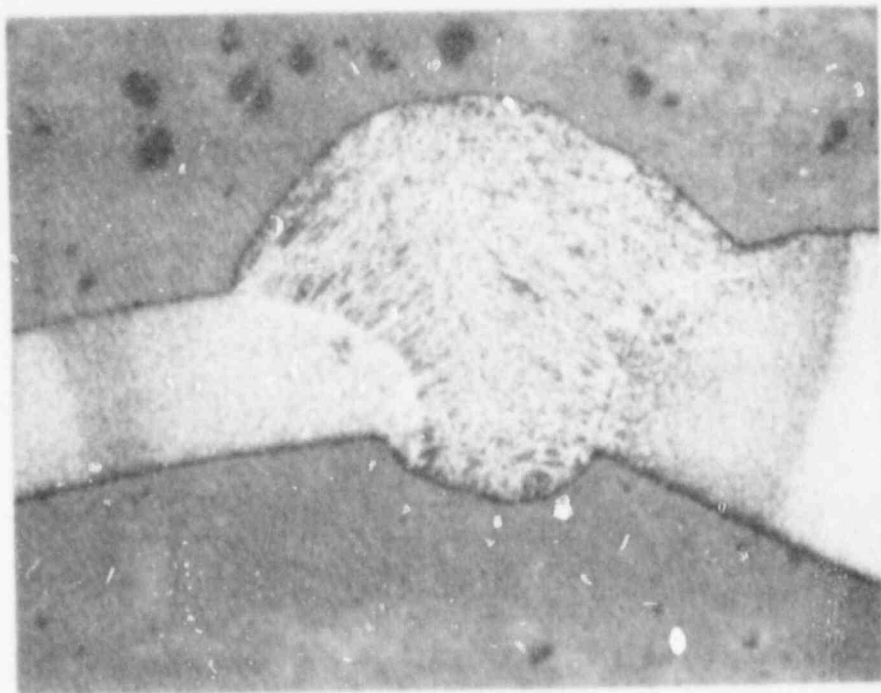


Figure 35 Cross-section of weld between cylinder  
and insert plate (horizontal position  
13x)

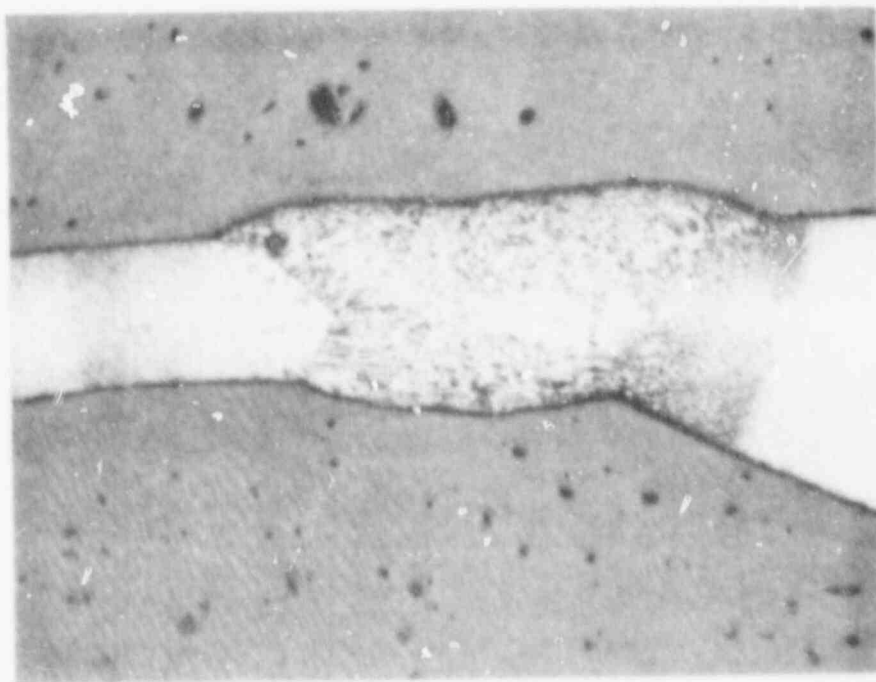


Figure 36 Cross-section of weld between cylinder  
and insert plate (vertical position 12x)

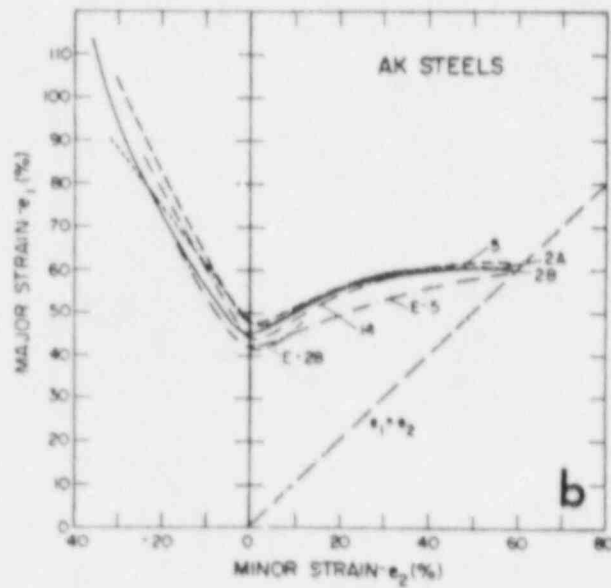
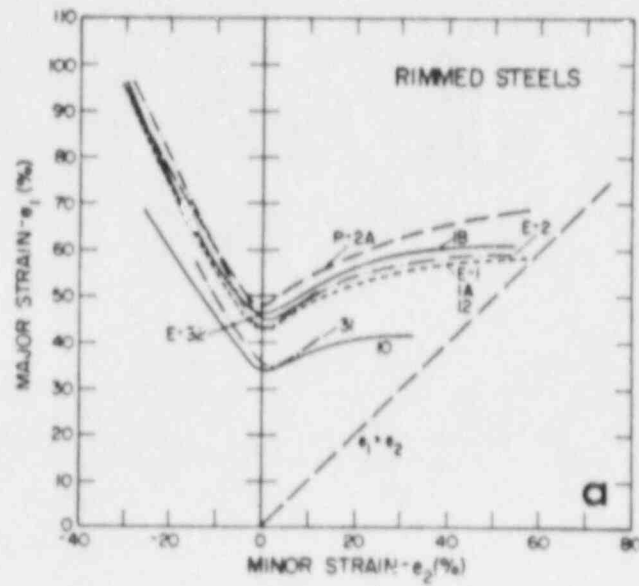


Figure 37 a) Forming limit curves for nine grades and lots of rimmed, low-carbon steels, b) Forming limit curves for six grades and lots of aluminum-killed (AK), low-carbon steels after Hecker (Ref. 3)



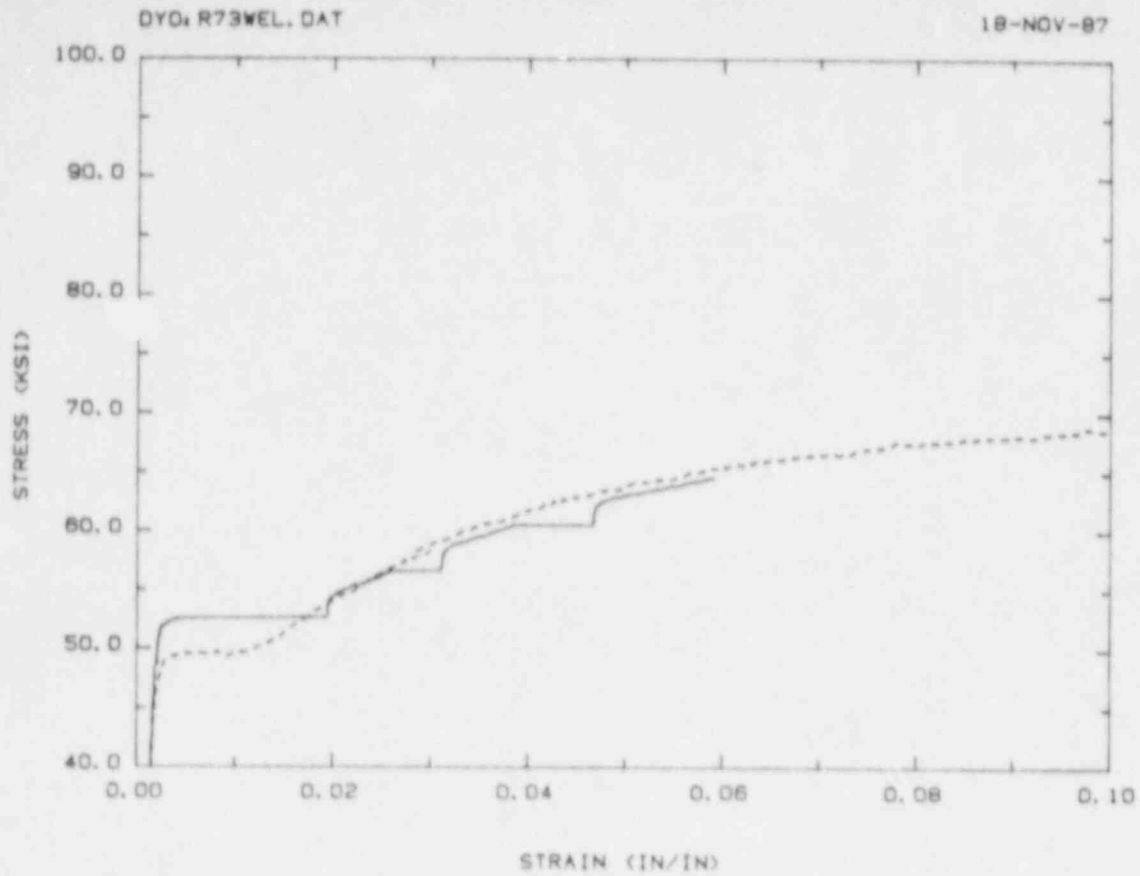


Figure 38 Comparison of monotonic and programmed load tensile behavior for cylinder material

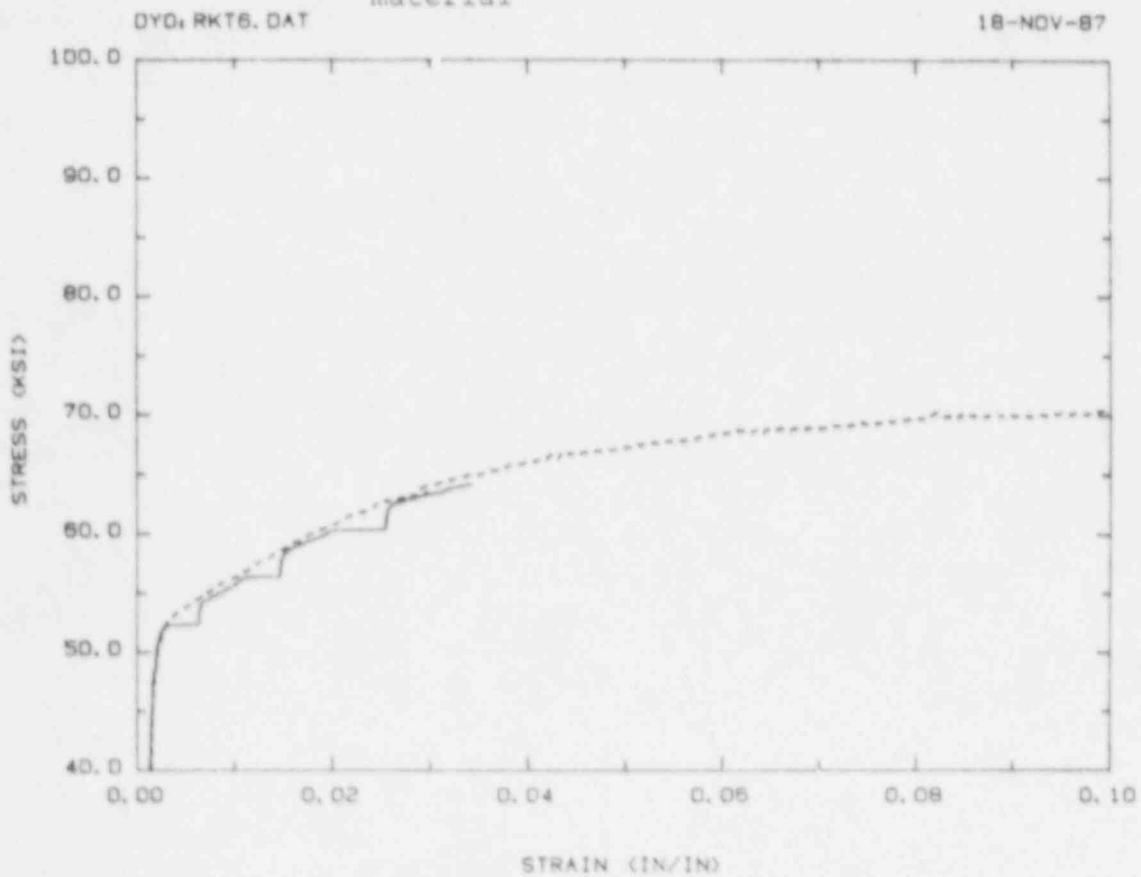


Figure 39 Comparison of monotonic and programmed load tensile behavior for dome material

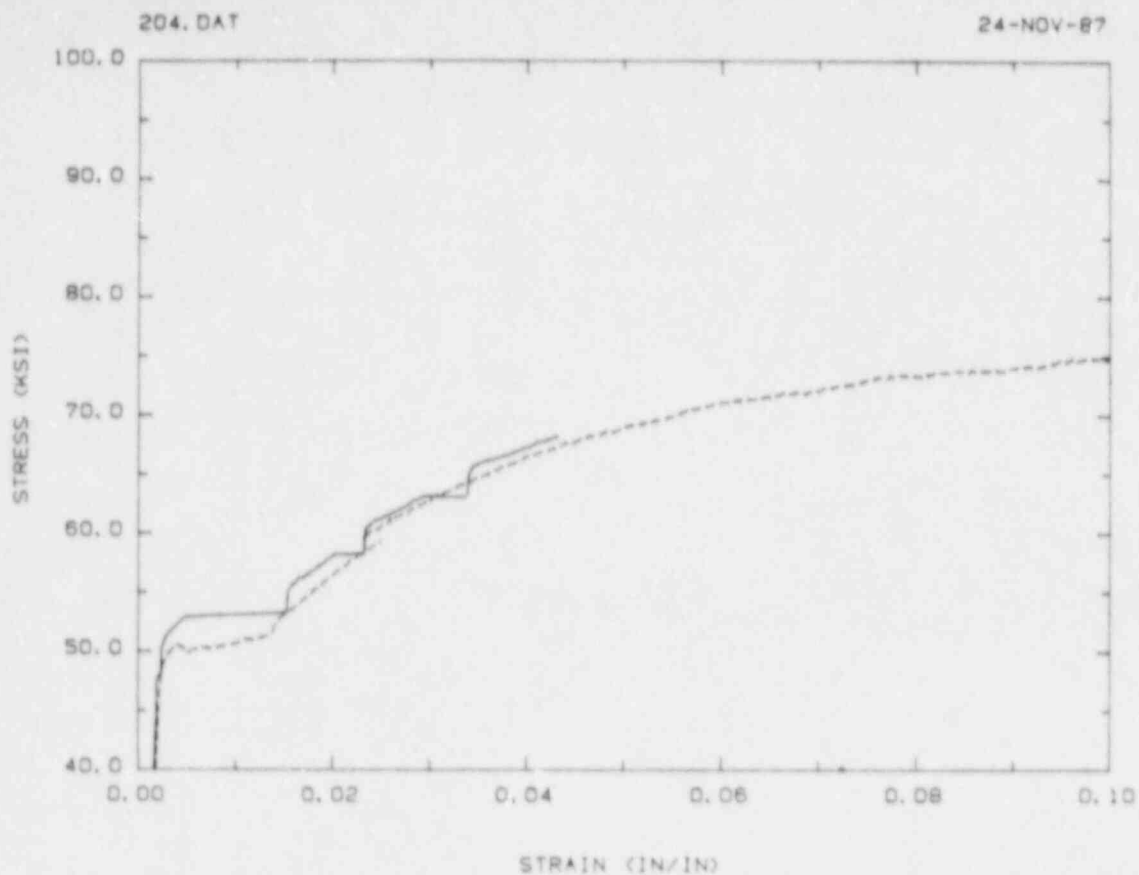


Figure 40 Comparison of monotonic and programmed load tensile behavior for insert plate material

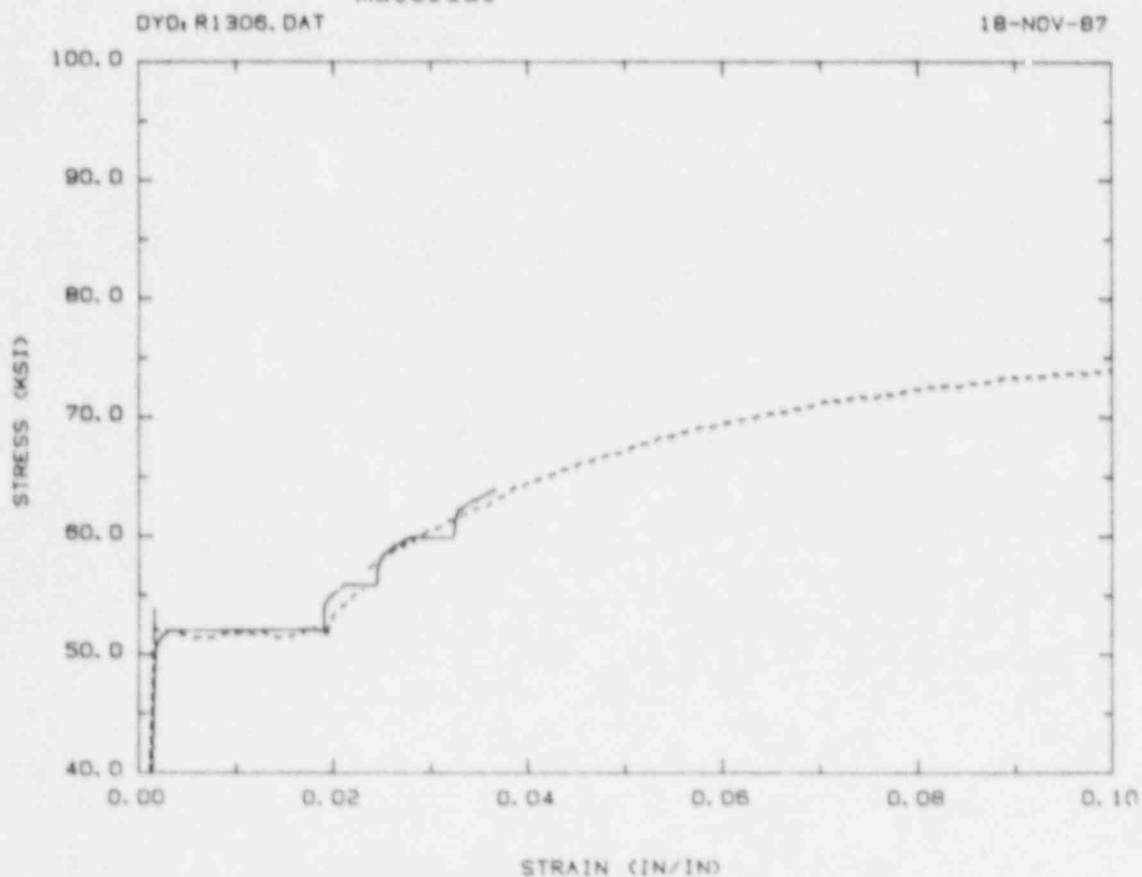


Figure 41 Comparison of monotonic and programmed load tensile behavior for penetration sleeve material 40

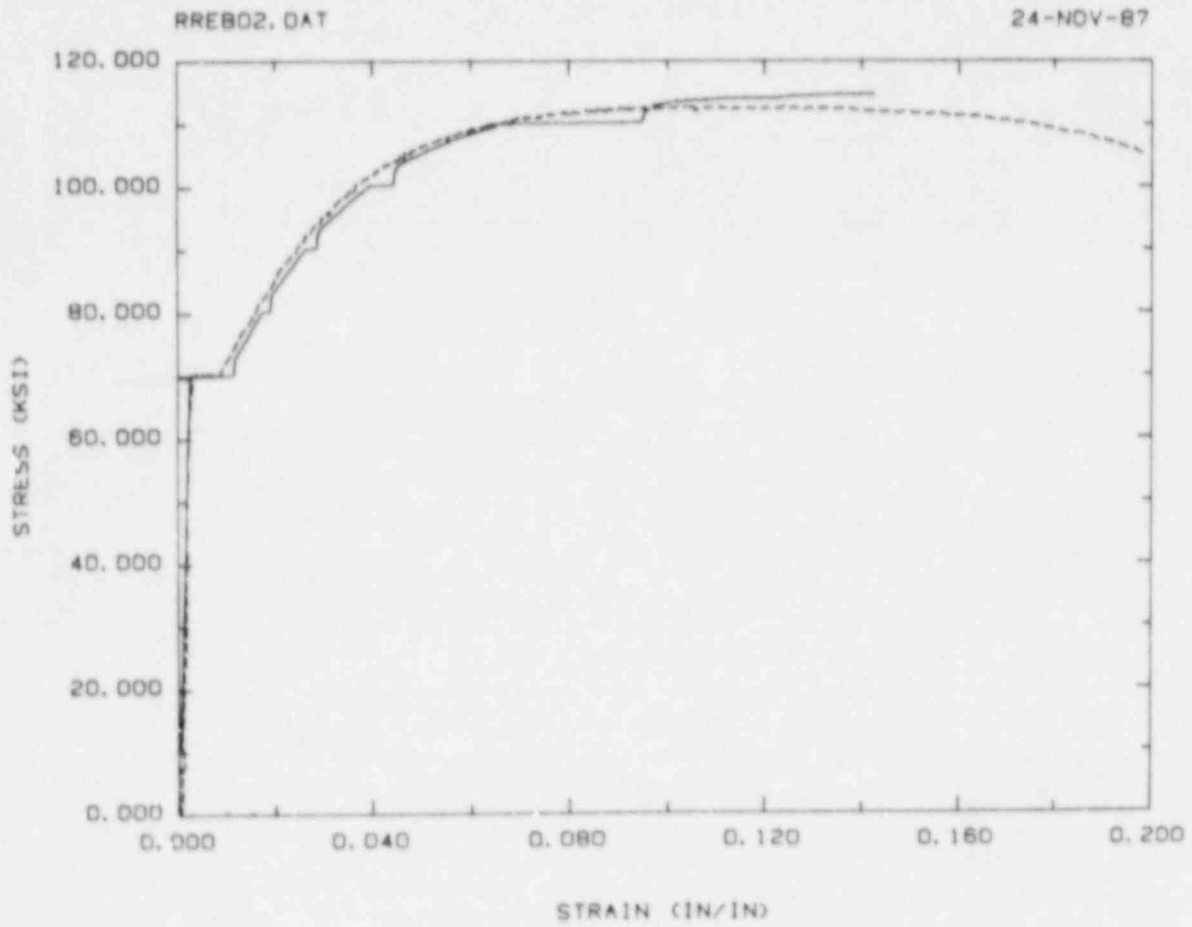


Figure 42 Comparison of monotonic and programmed load tensile behavior for #4 rebar

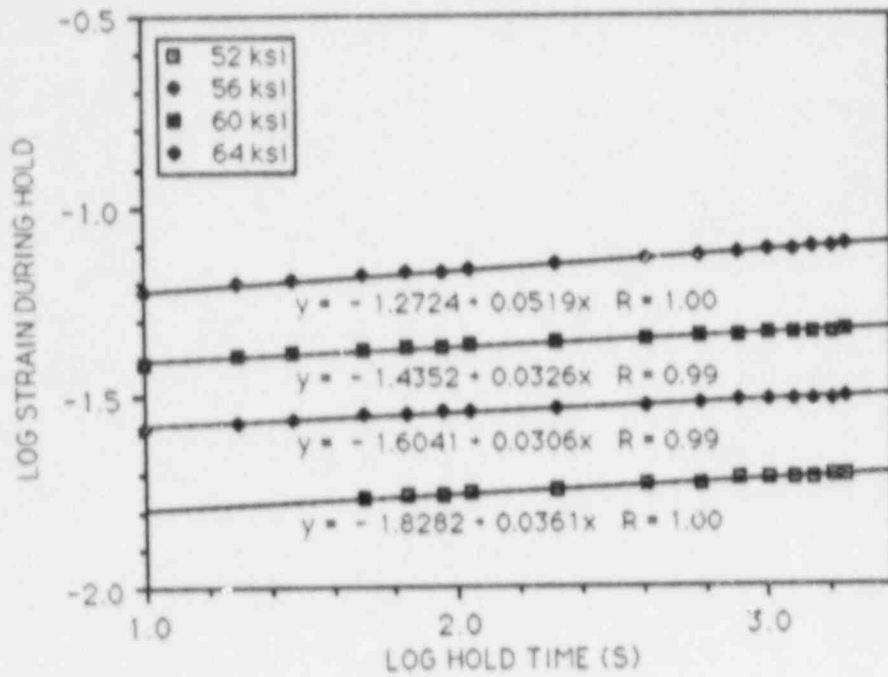


Figure 43 Creep behavior of cylinder material

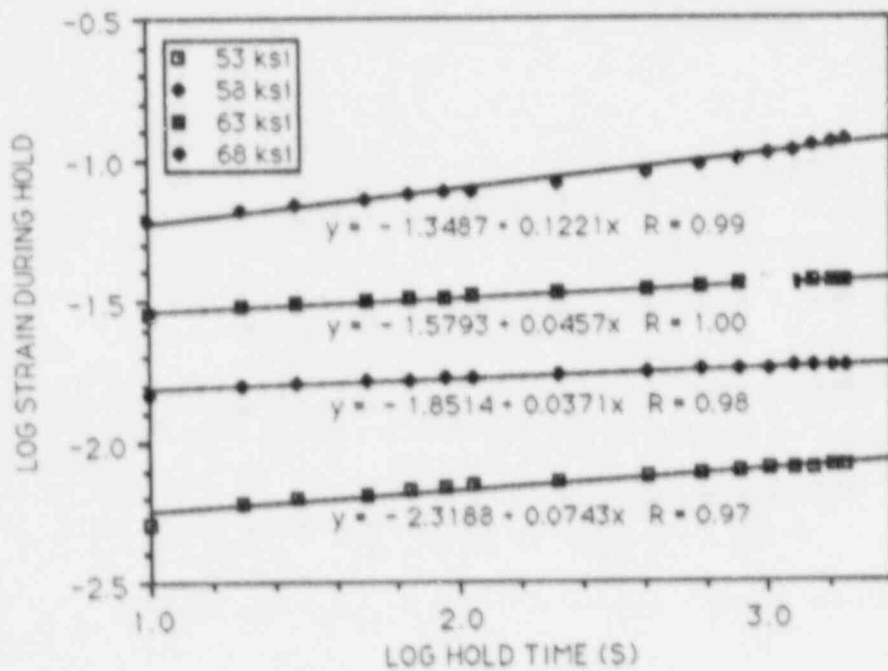


Figure 44 Creep behavior of dome material

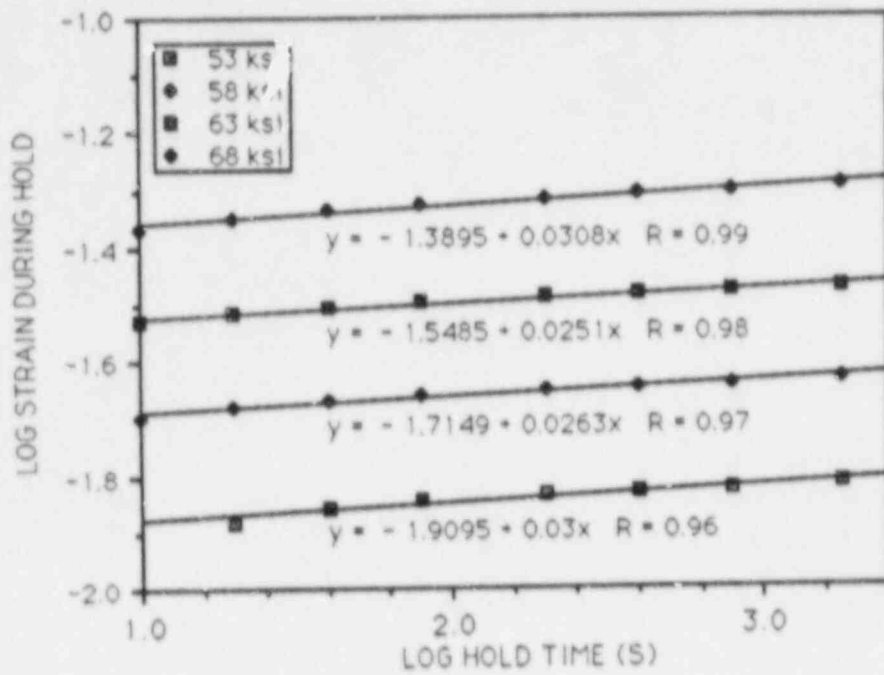


Figure 45 Creep behavior of insert plate material

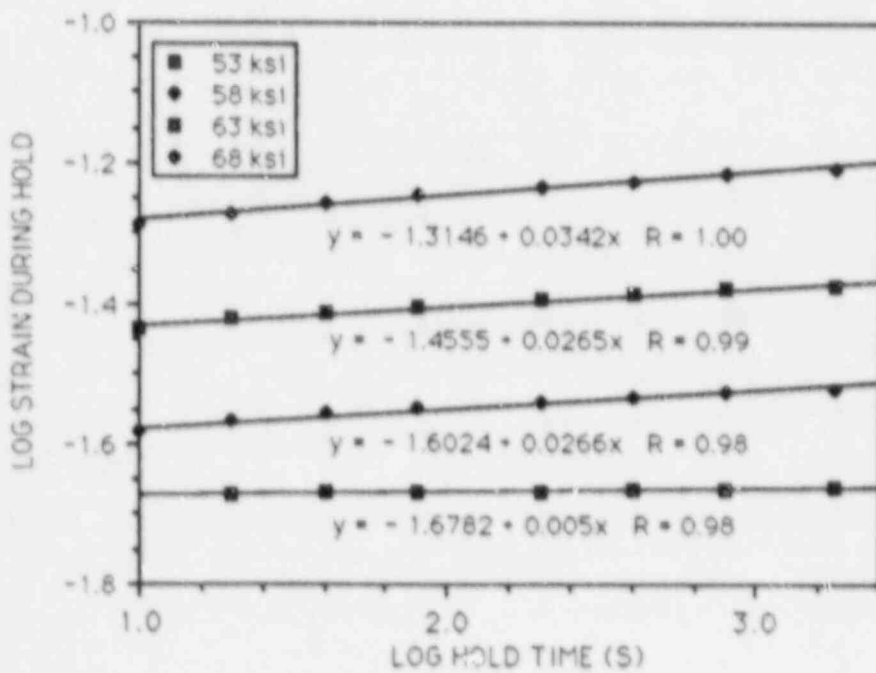


Figure 46 Creep behavior of penetration sleeve material

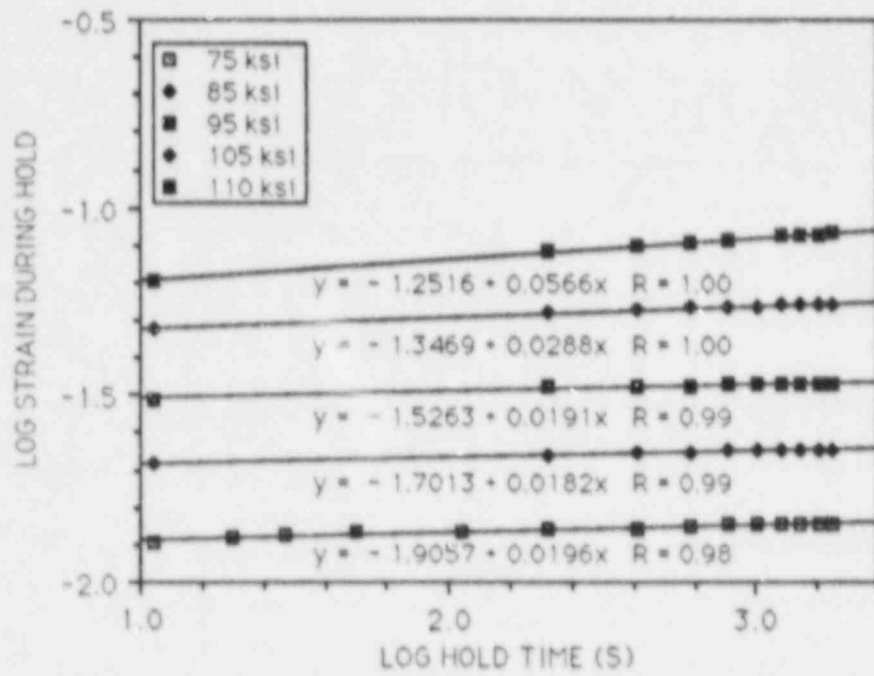


Figure 47 Creep behavior of #4 rebar

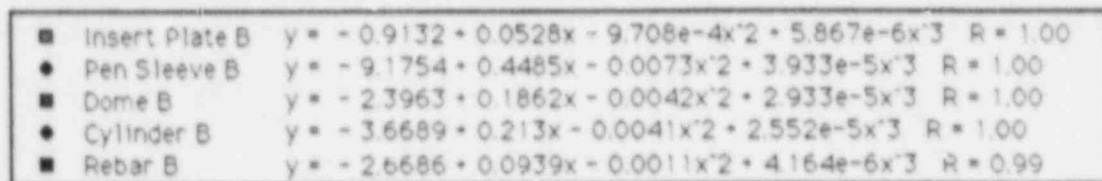
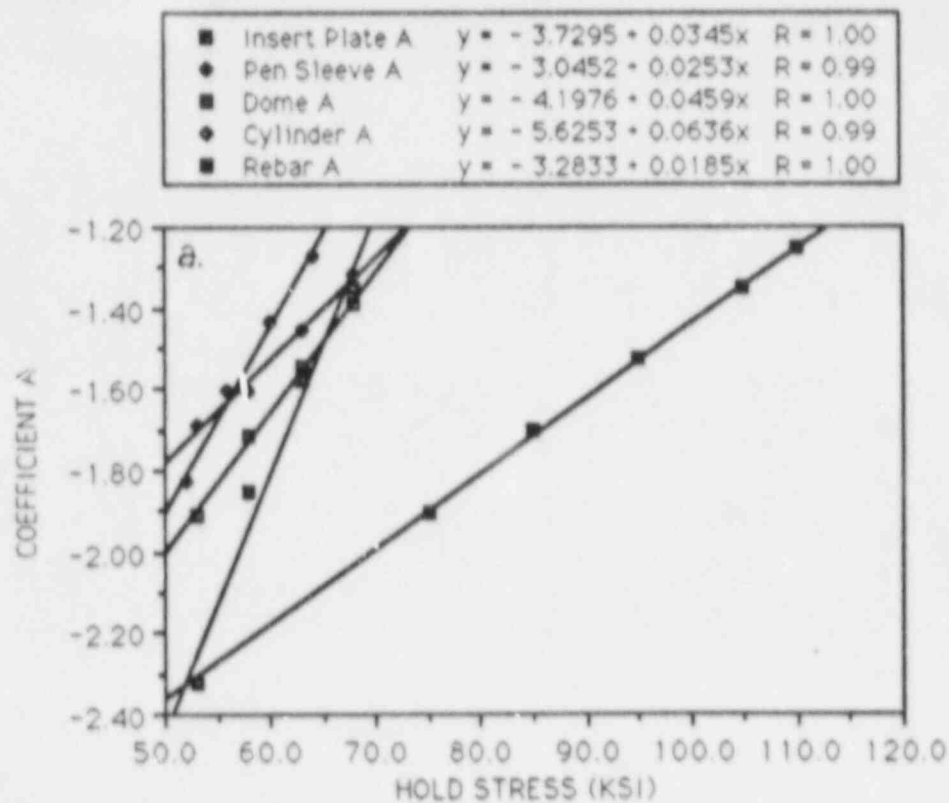


Figure 48 Compilation of creep equation coefficients:  
 a) Coefficient A,  
 b) Coefficient B

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13 ABSTRACT (200 words or less) This report summarizes the chemical analysis, metallography, and tensile test results obtained from steel materials and welds used in the construction of a 1/6-scale model reinforced concrete nuclear reactor containment. The purpose of building such a model is to experimentally verify the ability of numerical models to predict deformation and failure in full-size containments. Naturally, the predictions of such models are strongly influenced by the constitutive models used for the containment materials. The program reported on here was intended to provide such data. Besides providing tensile test data on the sheet, plate, and rebar materials used for dome, cylinder, cylinder inserts, reinforcements, & penetrations, pull tests on several weld geometries have been obtained. Standard tensile test data derived from load vs. extensometer output records (i.e. engineering stress vs. engineering strain) are supplemented with true stress vs. true strain data collected by measuring actual cross sectional areas during interrupted tensile tests. Additionally, "r" value measurements (ratio of width to thickness strain) which can be used to provide information about anisotropic multiaxial flow conditions are derived from the true strain data and reported. A small number of tests which included holds at constant load levels were also performed and analyzed to yield ambient temperature creep equations. Finally, conditions resulting from construction practices (such as residual stresses and the Bauschinger effect) which may affect the prediction of yield and fracture stresses from this data are briefly discussed.					
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