STONE & WEBSTER ENGINEERING CORPORATION

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REVISION STATUS

REVISION O

ORIGINAL ISSUE

REVISION 1

PAGES REVISED: 13

PAGES ADDED : N/A

PAGES VOIDED : N/A

REASON FOR CHANGE : DELETED CODE SUBPARAGRAPH REFERENCE

IN CONCLUSION NOTE 3 b.

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STRESS INTENSIFICATION FACTORS (SIF)

FOR UNREINFORCED LATERAL BRANCH CONNECTIONS

1.0 OBJECTIVE:

Paragraph NC 3650 of the ASME Section III, (12)*, code provides a method of analysis of class 2 and 3 piping products. The analysis utilizes Stress Intensification Factors (SIF) along with some simple equations to establish acceptable limits on stresses.

Figure NC 3673.2(b)-1, (12)*, provides SIF's for commonly used standard piping products. For non-standard piping products NC 3673.2(b), (12)*, states that SIF may be taken as $C_2K_2/2$ where C_2 and K_2 are class 1 stress indices given in Table NB 3682.2-1, (12)*. Lateral branch connections are not among the components for which SIF or C_2 and K_2 stress indices are available. Thus, code evaluation of lateral branch connection by simplified analysis is not possible.

The purpose of this calculation is to establish a method to derive an appropriate SIF for use in the evaluation of lateral "stub-in" branches using NC-3650 rules, (12)*, based upon the results of a published literature survey.

^{*} Numbers in () indicate reference numbers listed in Section 3.

0.7, 0.8, 0.9, 0.95, 1.0

indicates increments will be used applying to 70%, 80%, 90%, 95%, and 100% of the load history (as defined on the loading cards).

- 4. TIMEP is the period of this analysis step. Default value is one which means that normalized time is used.

 Note: this time period for static analysis is not accumulated over different steps. The time period must be non-zero.
- 5. NUMBER this parameter is used to initiate the automatic load incrementation option and to suggest the increment size. For example, NUMBER = 5 will make the first increment equal to 0.2 of the total time. However, the program will now adjust the time increment based on the number of cycles needed in each increment with the limitation being that the time increments will not be larger than 0.2 of the total time.

Note: NUMBER and DIRECT are mutually exclusive parameters.

6. CUTMAX - the maximum number to times the suggested uniform increment size may be subdivided.

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2.0 ASSUMPTIONS 1. It is reasonable to assume that unreinforced lateral branch connection behavior would also be indicative of reinforced lateral branch connection behavior. 2. Bending moment stress indices (c, and k,) are same for tubular joints (no hole at the intersection) as well as pipe branch connections.	7	PAGE.	SK CODE	OPTIONAL T		P			
1. It is reasonable to assume that unreinforced lateral branch connection behavior would also be indicative of reinforced lateral branch connection behavior. 2. Bending moment stress indices (c ₂ and k ₃) are same for tubular joints (no hole at the intersection) as well as pipe branch connections.		CONFIRM REQUIRE				SUMP			
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- 12. ASME Boiler and Pressure Vessel Code, Section III, 1980 Edition.
- 13. ANST B31-1, American National Standard Code for Pressure Piping, Power Piping, 1977 Edition.

4.0 ANALYSIS

Stress intensification factors (i) for various piping components are, in large part, based upon Markl's (1)* approach and test data generated by his work.

The Markl cyclic moment fatigue tests did not cover lateral branch connections and code does not provide either 'i' factors or \mathcal{C}_2 , K_2 stress indicies for lateral branch connections. Several authors have investigated this problem and references $(2)^*$, $(3)^*$, and $(4)^*$ are some of the significant published reports.

WRC Bulletin 200 (2)* reports and analyzes test data on branch connections including unreinforced laterals exposed to internal pressure and/or external couples. Tests wer- conducted on plain pipes, tees (normal branch connections) and 45° laterals, to determine "limit loads" of the components when subject to internal pressure, in-plane and out-of-plane bending moments. No fatigue tests were conducted.

Figures 1 and 2, (2)*, show results of in-plane couple tests on a 45° lateral and a 90° normal branch connection, both of identical dimensions except for the angle of branch connection. Table 1, (2)*, summarizes the data and test results as well. From these figures and table, it is clear that the limit load for the lateral is higher compared to that of 90° tee indicating that the laterals are plastically stronger than tees. The same conclusions are arrived at for out-of-plane couples as observed by test results shown in figures 3, 4, and 5 and Table 2, (2)*. However, as observed in the Pe/P₁ column in Table 1, (2)*, the following is concluded:

- Laterals are plastically stronger than 90° branch connections when subjected to external bending moments.
- Laterals are plastically weaker compared to tees when exposed to internal pressures.

Conclusion 2 is drawn as a point of interest since in class 2, 3, and B31.1 stress analysis, the pressure loading terms are not intensified. The branch construction need only meet the local pressure reinforcement requirements stated in the applicable codes.

While reference (2)* conducted tests on 45° lateral branch connections only reference (3)* reports results of tests conducted on tubular joints laterals with angle of branch connections ranging from 0° to 90°. Reference (3)* presents the finite element model analytical results as well as results of experimental investigations and provides stress concentration factor (SCF) for radial thrusts loads, in-plane, out-of-plane bending moments applied to both normal and lateral branch connections. The SCF value reported does not represent maximum or peak stress in the joint, but rather the equivalent of the maximum primary plus secondary membrane plus bending stresses as represented by the C2 stress indices of NB 3650.

4. # 214

B2-12241-9

^{*} Numbers in () indicate reference numbers listed in Section 3.

^{##} FIGS. \$TABLES POLLEN CONCLUSION (ELTION

J.G. Kuang, et al, (3)* performed a parametric study to best fit the experimental results to those arrived at by finite element analysis. The following parameters that govern the stress distribution in branch connections were chosen: the run thickness to diameter ratio T/D, the branch the run pipe thickness ration t/T, the branch to run diameter ratio d/D, and most significantly to our current investigation, the angle of branch connection. The T/D and t/T ratios govern the stress distribution by influencing the radial flexibility of the run pipe (Figure 6), (3)*, and by bending stress in the branch at the intersection. The d/D ratio and the angle of branch connection influence the stress distribution by the load transfer mechanism.

when a tee is subjected to axial branch loading, the load is transferred to the run primarily via local bending and punching shear. As a result if the branch is inclined at other than 90° only the component of the load normal to the run wall is of primary concern, as the horizontal component is transferred by compression or tension in the run.

Reference (3)* developes empirical equations to derive SCF's based on the parameters discussed a ove. The applicability and accuracy of the empirical equations developed were verified by comparing the test results to results obtained by the empirical equations. Figure 7(3)* shows good agreement between the experimental and analytical results. The empirical expression for radial loads, in-plane, and out-of-plane bending loads contain a multiplying term $\sin^3\theta$ where θ is the angle. of branch connection, [see Table 3] and 'a' is a constant greater than zero. This indicates that the maximum stress for the branch connection and SCF increases as the angle 0 increases, 0°< 0 <90°.

Several other researchers' investigations, (6)*, (7)*, and (8)* have resulted in empirical equations for SCF's that include a sin 0 term reinforcing the conclusion arrived at in reference (3)*. Table 4, (9)*, also indicates the agreement between the results arrived at by various authors.

Reference (3)* imposes a set of limitations on geometric parameters to minimize dispersion between experimental and analytical results. Most important of them are:

- $0.2 \le t/T \le 0.8$
- 0.3 \(d/D \(\) 0.8
- 0° ≤ 0 ≤ 90°

As mentioned earilier, the SCF derived in reference (3)* is akin to C2 stress indices of the code. From reference (3)* it is clear that the SCF and therefore, C2 for laterals are lower thean 90° branch connections. By considering the geometry, branch and lateral connections must have similar K2 indices and thus the product of C2K2 for laterals will always be lower than that for branches. Therefore stress intensification factor 'i' which is equal to $C_2R_2/2$ (NC 3673.1(b)) for laterals will always be lower than that for 90° branches.

^{*} Numbers in () indicate reference numbers listed in Section 3.

While it is true that Reference (3)* deals with tubular joints, (no hole at the run-branch intersection) the same conclusions are arrived at by C. Hsiao and A.S. Kahn (4)* for piping intersetions. C. Hsiao, et al, present the result of finite element investigation of 6 x 3 branch connections for three branch angles, namely 30°, 60°, and 90°, subjected to in-plane and out-of-plane loads.

The output of the finite element analysis in the form of membrane stresses and bending moments in the local coordinates are properly combined to obtain total stresses at the run/branch pipe intersection. These stresses are divided by "beam type stresses" (= M/Zr) for the known applied in-plane and out-of-plane moment to obtain a "stress ratio". Here M is the applied moment and Zr the elastic section - modules of the run pipe. The stress ratio thus obtained are compared to the experimental data, (10)*, for in-plane and out-of-plane loadings available for 30° and 60° branch connection angles. Figures 8, 9, 10, and 11, (4)*, exhibit good agreement between the experimental and analytical results. Reviewing figures 12 through 20, (4)*, indicate that the stress ratio increases with the increase in the angle of branch connection thus confirming results of reference (3)*.

Reference (11)* provides a basis to evaluate the fatigue capacities of a lateral in comparison to a 90° branch of equal size under equivalent loading conditions. Figure 21 results indicate a trend to initial failure (crack initiation) occurring at 90° branches sooner than at equivalent laterals and complete failure (through-wall crack) predominantly occurring in 90° branch connections. The tests were run using stress ranges above those allowed by the piping codes for a 7,000 cycle life. The laterals withstood greater than 7,000cycles in all cases. These trends indicate that a lateral is less sensitive to cyclic damage than its equivalent 90° branch.

^{*} Numbers in () indicate reference numbers listed in Section 3.

5.0 SUMMARY OF RESULTS

- Under a single load to failure, the lateral branch connection is stronger plastically than the normal branch connection.
- Under cyclic loading the lateral branch connection has greater fatigue strength than a normal branch connection.
- Laterals are plastically weaker compared to tees when exposed to internal pressure.
- 4. While reference (3)* tests are on tubular joints the results are applicable to branch connections considering the similar conclusions arrived at in reference (1)* and (4)*.
- 5. Reference (3)* imposes geometric parameter limitations for the empirical expressions for SCF to be valid. However, it is reasonable to assume that the dispersion between experimental and analytical results for tees and laterals will be in the same direction so that the laterals SCF's will always be lower than that of tees even outside the geometric limitations.

^{*} Numbers in () indicate reference numbers listed in Section 3.

6.0 CONCLUSIONS

- 1. In a piping analysis a lateral branch connection can be safely analyzed inclar to a normal branch connection for class 2 and 3 and non-A. Spes.
- In general the results of this investigation indicates that unreinforced fabricated lateral connections are stronger than corresponding 90° connections. It is expected that this same conclusion could be drawn for reinfored lateral connections as well. However, until more data becomes available to confirm this it is recommended that a factor be applied for additional conservatism in the calculation of stress intensification factors (SIF) for laterial connections.

The SIF for either an unreinforced or reinforced connections may be determined by calculating the SIF for an equivalent 90° connection and increasing it by 25 percent.

- 3. The above must be subjected to the following restrictions:
 - a. The branch connections must meet all the pressure reinforcement requirements of NC 3643 of ASME Section III or ANSI B31.1 Paragraph 104.3.1.
 - b. The branch connection angle (the angle between the run and branch pipe axes) is between 45° and 90° both inclusive

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- c. ASME Section III code version is limited to the summer 1981 addenda and earlier. Winter 1981 addenda requires that pressure terms be modified by "B" indices.
- 4. The use of lateral branch connections would be restricted to ASME Class 2 and 3 and B31.1 systems; insufficient data exists to draw conclusions for Class 1 applications.

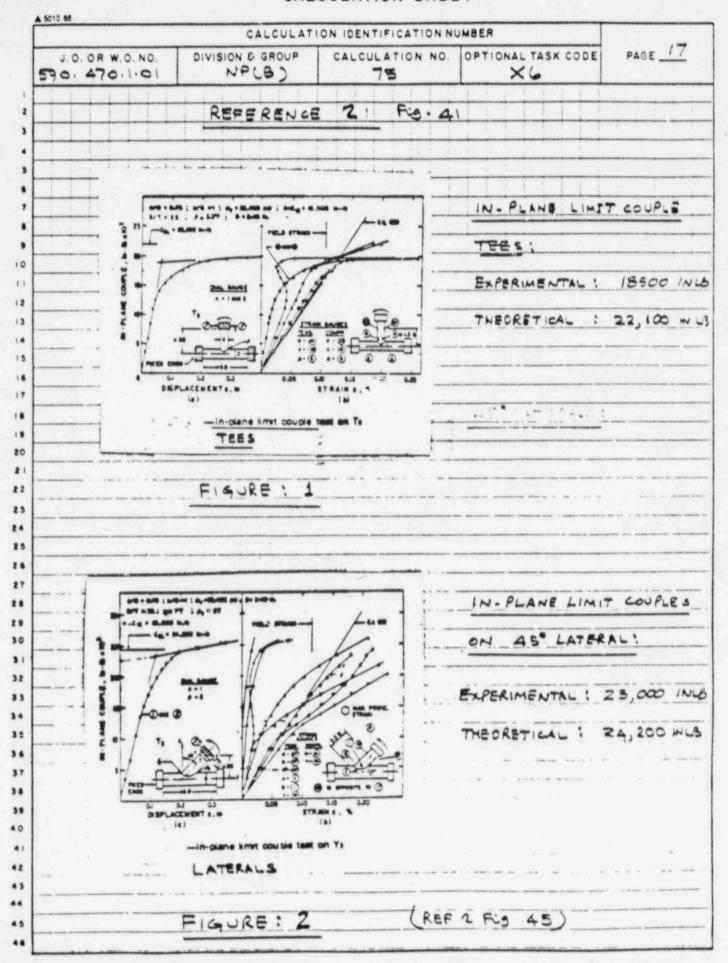
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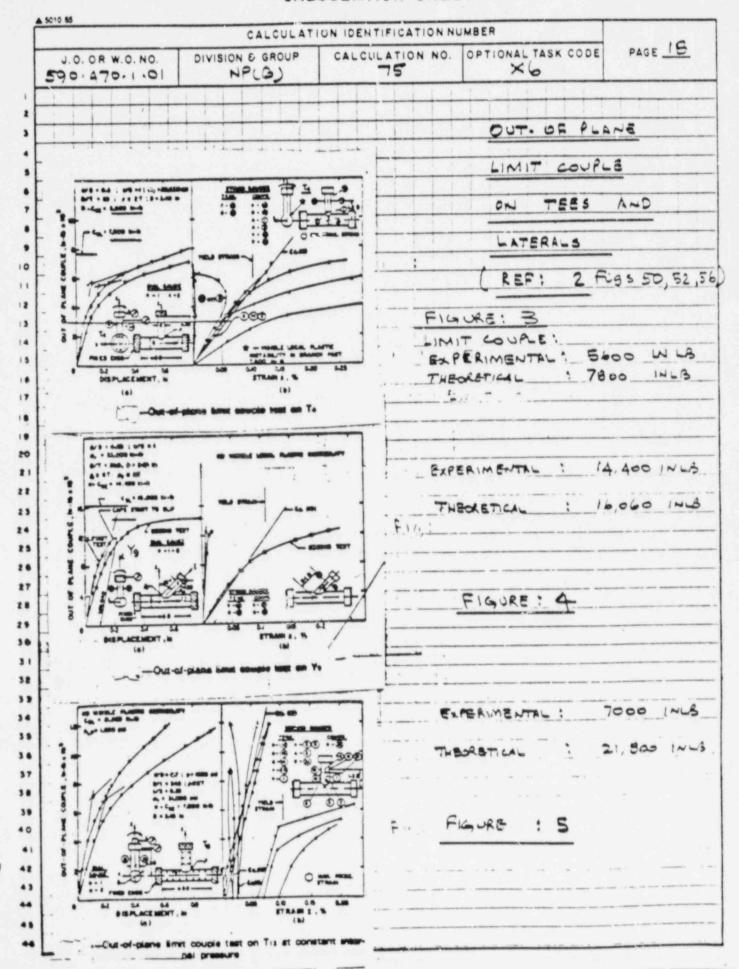
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TABLE 3 (REF 3 TABLES)

STRESS CONCENTRATION FACTORS FROM SPESCIZ, OTC 2205 AUG, 197 T, Young = 1.177 (To) -0.608 e -1.2 (40) (47) 1.333 (02) -0.057 sin 1.494 & T T, Young = 2.784 (To) -0.55 e -1.35 (40) (47) (02) -0.12 sin 1.94 & T T, Young = 0.949 (To) -0.666 (do) -0.059 (tr) 1.104 (90) -0.07 sin 1.521 & T Kong = 0.825 (To) -0.157 (do) -0.441 (tr) -0.560 (90) -0.058 (1.400 sin 0) T, Kong = 1.26 (To) -0.56 (do) -0.36 (tr) -0.68 (9.492) 0.126 sin 0.50 T, Kong = 1.26 (To) -0.16 (do) -0.36 (tr) -0.68 (9.492) 0.126 sin 0.50 T, Kong = 1.288 (To) -0.1 (do) -0.36 (tr) -0.68 (9.492) 0.126 sin 0.50 T, Kong = 0.463 (To) -0.123 (do) -0.36 (tr) -0.68 (9.492) 0.126 sin 0.50 T, Young = 0.463 (To) -0.60 (do) -0.06 (tr) -0.68 sin 0.57 & T T, Young = 0.463 (To) -0.35 (do) -0.06 (tr) -0.94 sin 0.94 Kong = 1.40 (To) -0.38 (do) -0.06 (tr) -0.94 sin 0.94 T, Young = 0.507 (To) -0.04 (do) -0.05 (tr) -0.989 sin 1.557 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.889 sin 1.557 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.649 sin 0.543 sin 2.033 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.643 sin 2.033 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.643 sin 2.033 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.643 sin 2.033 & T T, Young = 0.229 (Tr) -1.04 (do) -0.619 (tr) -0.643 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.543 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.543 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.543 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.543 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.543 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 & T T, Young = 0.83 (To) -0.852 (do) -0.801 (tr) -0.843 sin 2.033 &	J.O. OR W.O. N 590-470-1		15
T, Y branch = 2.78+(T ₀) -0.55 e ^{-1.35} (d ₀) ³ (t/7)(D ₁) -0.12 sin 1.94 to T, Y branch = 0.949(T ₀) -0.6666 (d ₀) -0.059(t/7) 1.104 (d ₀) 0.067 sin 1.521 to K branch = 0.825(T ₀) -0.157 (d ₀) -0.441 (t/7) -0.560 (d ₀) 0.056 e ^{-1.660} sin to T, K chard = 1.26(T ₀) -0.54 (d ₀) -0.12 (t/7) 1.068 sin to T, K stanch = 5.65(T ₀) -0.1 (d ₀) -0.36 (t/7) 1.068 (d ₁ +q ₂) 0.126 sin 0.5 to 0 < t < 4.5° T, K stanch = 12.88(T ₀) -0.1 (d ₀) -0.36 (t/7) 1.68 (d ₁ +q ₂) 0.126 sin 0.5 to 45° < t < 90° Tot T, K = 4.491(T ₀) -0.123 (d ₀) -0.396 (t/7) 0.68 (d ₁ +q ₂) 0.159 sin 1.267 to T, Y chard = 0.463(T ₀) -0.6 (d ₀) -0.06 (t/7) 0.86 sin 0.57 to T, Y stanch = 1.40(T ₀) -0.38 (d ₀) 0.06 (t/7) 0.94 sin 0.9 to T, Y chard = 0.507(T ₀) -0.38 (d ₀) 0.06 (t/7) 0.94 sin 0.9 to T, Y chard = 0.507(T ₀) -0.04 (d ₀) 0.06 (t/7) 0.889 sin 1.557 to 0.3 < d/0 < 0.55 T, Y chard = 0.229(T ₀) -1.014 (d ₀) -0.801 (t/7) 0.543 sin 2.033 to 0.3 < d/0 < 0.55 T, Y stanch = 0.843(T ₀) -0.852 (d ₀) -0.801 (t/7) 0.543 sin 2.033 to T. Bandyerse Bending = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281 (t/7) 0.543 sin 2.033 to T. Branch = 0.441 (T ₀) -0.852 (d ₀) -0.281			197
Typeracle 2.784 (70) 0.55 e 1.35 (40) (4/1) (D) 0.12 sin 1.94 to The sin 1.94 to Typeracle 2.784 (70) 0.666 (4/1) 0.059 (1/1) 1.104 (9/1) 0.067 sin 1.521 to Typeracle 2.825 (70) 0.157 (4/1) 0.560 (9/1) 0.067 sin 1.521 to Typeracle 2.825 (70) 0.157 (4/1) 0.68 (9/1) 0.056 e 1.660 sin to Typeracle 2.86 (70) 0.157 (4/1) 0.68 (9/1) 0.126 sin to Typeracle 2.88 (70) 0.14 (1/1) 0.68 (9/1) 0.126 sin 0.5 to 0 < to 4 45° T.K. Strank = 12.88 (70) 0.14 (1/1) 0.36 (1/1) 0.68 (9/1) 0.126 sin 0.5 to 0 < to 4 45° T.K. Strank = 12.88 (70) 0.123 (1/1) 0.38 (1/1) 0.68 (9/1) 0.126 sin 0.5 to 0 < to 4 45° T.Y. Chord = 0.463 (70) 0.123 (1/1) 0.124 (1/1) 0.124 sin 0.5 to 0 < to 4 45° T.Y. Chord = 0.463 (70) 0.123 (1/1) 0.124 (1/1) 0.125 sin 0.12 to 0 < to 4 45° T.Y. Chord = 0.463 (70) 0.123 (1/1) 0.124 sin 0.12 to 0 < to 4 45° T.Y. Chord = 0.463 (70) 0.124 (1/1) 0.125 sin 0.12 to 0 < to 4 45° T.Y. Chord = 0.507 (70) 0.104 (1/1) 0.125 sin 0.12 to 0 < to 4 45° T.Y. Chord = 0.507 (70) 0.104 (1/1) 0.125 sin 0.12 to 0 < to 4 45° T.Y. Chord = 0.229 (7/1) 0.104 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1/1) 0.125 sin 0.157 to 0 < to 5 5 6 (1	1	T, Ychan = 1,177 (To) -0808 e-1.2(%) (t/) 1.353 (0/) -0.057 sin 1.694 &	7
Colored = 0.949 (To) -0.666 (do) -0.059 (to) -0.058 (1.104 (go) 0.058 (1.440 sin to to) -0.825 (To) -0.157 (do) -0.441 (to) -0.560 (go) -0.058 (1.440 sin to to) -0.566 (do) -0.056 (do) -0.057 (do) -0.057 (do) -0.056 (do)			T
K Breach = 0.825 (Tp) -0.157 (dp) -0.441 (t/T) -0.560 (9p) 0.056 2 1.440 sin = T. T.K Chard = 1.26 (Tp) -0.54 (dp) -0.35 (t/T) -0.68 sin = T. T.K Branch = 5.65 (Tp) -0.1 (dp) -0.35 (t/T) -0.68 (9.492) 0.126 sin 0.50 = 0.56 (45) -0.126 (47) 0.68 (9.492) 0.126 sin 0.50 = 0.56 (47) 0.68 (9.492) 0.126 sin 0.50 = 0.56 (47) 0.68 (9.492) 0.126 sin 0.50 = 0.56 (47) 0.672 (9.492) 0.159 sin 0.20 = 0.45 (Tp) -0.123 (dp) -0.36 (t/T) 0.86 sin 0.57 = T. T.Y Chard = 0.463 (Tp) -0.6 (dp) -0.06 (t/T) 0.86 sin 0.57 = T. T.Y Chard = 1.40 (Tp) -0.38 (dp) -0.38 (t/T) 0.38 sin 0.21 = 0.50 (Tp) -0.38 (dp) -0.35 (t/T) 0.38 sin 0.9 = 0.50 (Tp) -0.35 (t/T) 0.35 sin 0.5 = 0.55 (t/T) 0.35 sin 0.57 = 0.55 (dp) -0.55 (t/T) 0.35 sin 0.57 = 0.55 (dp) -0.55 (t/T) 0.35 sin 0.57 = 0.55 (dp) -0.55 (dp) -0.35 (t/T) 0.35 sin 0.57 = 0.55 (dp) -0.55	797/		T
T, K _{chord} = 1.26 (T _D) ^{-0.54} (d _D) ^{0.12} (t _T) ^{1.068} sin \$\phi\$ T, K _{stranch} = 5.65 (T _D) ^{-0.1} (d _D) ^{-0.36} (t _T) ^{0.68} (\frac{9.+9.2}{D})^{0.126} sin 0.5\$\phi\$ O < \$\phi\$ & 4.5* T, K _{stranch} = 12.88 (T _D) ^{-0.1} (d _D) ^{-0.36} (t _T) ^{0.68} (\frac{9.+9.2}{D})^{0.126} sin 0.5\$\phi\$ A5° < \$\phi\$ & 90° Tof T, K = 4.491 (T _D) ^{-0.423} (d _D) ^{-0.396} (t _T) ^{0.692} (\frac{9.+9.2}{D})^{0.159} sin 0.57\$\phi\$ T, Y _{chord} = 0.463 (T _D) ^{-0.6} (d _D) ^{-0.06} (t _T) ^{0.86} sin 0.57\$\phi\$ T-1 K _{chord} = 1.40 (T _D) ^{-0.23} (d _D) ^{-0.38} (t _T) ^{0.38} sin 0.21\$\phi\$ T-1 K _{chord} = 1.40 (T _D) ^{-0.38} (d _D) ^{0.06} (t _T) ^{0.94} sin 0.9\$\phi\$ T-1 K _{chord} = 1.40 (T _D) ^{-0.38} (t _T) ^{0.35} sin 0.5\$\phi\$ T-1 T, Y _{chord} = 0.507 (T _D) ^{-1.014} (d _D) ^{0.787} (t _T) ^{0.889} sin 1.557\$\phi\$ T-1 0.3 < d _D < 0.55 T, Y _{chord} = 0.229 (T _D) ^{-1.014} (d _D) ^{-0.892} (d _D) ^{0.801} (t _T) ^{0.543} sin 2.035\$\phi\$ T-1 0.55 \(\leq d_D \leq 0.75\) T, Y _{chord} = 0.843 (T _D) ^{-0.852} (d _D) ^{0.801} (t _T) ^{0.543} sin 2.035\$\phi\$ T-1 0.3 < d _D < 0.55 T, Y _{stranch} = 0.843 (T _D) ^{-0.852} (d _D) ^{-0.852} (d _D) ^{0.543} sin 2.035\$\phi\$ T-1 0.5 < d _D < 0.55	* 2 2		7
T. Kstanch = 5.65(Tp) -0.1 (dp) -0.36 (t/T) 0.68 (9,+92) 0.126 sin 0.50 O C O C O C O C O C O C O C O C O C O			-
T, Kbreich = 12.88 (To) -0.1 (do) -0.36 (to) -0.48 (gi+gi) 0.126 sin 2.88 (to) -0.123 (do) -0.396 (to) -0.123 (do) -0.396 (to) -0.123 (do) -0.124 (do) -0.125 (do) -0.124 (do) -0.125 (do) -0.124 (do) -0.125 (do)		T. Keranch = 5.45(To) -0.1(do) -0.76 (t/) 0.68 (9,+92) 0.126 sin 0.50	
Tof T, K = 4.491 (T/O) -0.223 (d/O) -0.36 (t/T) 0.80 sin 0.57 & T- T, Y Chard = 0.463 (T/O) -0.6 (d/O) -0.04 (t/T) 0.86 sin 0.57 & T- T, Y Brend = 1.109 (T/O) -0.23 (d/O) -0.38 (t/T) 0.38 sin 0.21 & T- K Chard = 1.40 (T/O) -0.38 (d/O) 0.06 (t/T) 0.94 sin 0.9 & T- K Branch = 2.877 (d/O) -0.35 (t/T) 0.35 sin 0.5 & T- T, Y Chard = 0.507 (T/O) -1.014 (d/O) -0.619 (t/T) 0.889 sin 1.557 & T- 0.3 < d/O < 0.55 T, Y Chard = 0.229 (T/O) -1.014 (d/O) -0.619 (t/T) 0.889 sin 1.557 & T- 0.55 & d/O & 0.75 T, Y Branch = 0.243 (T/O) -0.852 (d/O) 0.801 (t/T) 0.543 sin 2.033 & T- 0.3 < d/O < 0.55 T, Y Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T, Y Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T, Y Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. Branch = 0.441 (T/O) -0.852 (d/O) -0.281 (t/T) 0.543 sin 2.033 & T- T. B	Sira	T, Koranch = 12.88 (76) -01 (06) -0.36 (4/7) 0.68 (9,+92) 0.126 sin 2.886	Τ.
TY Branch = 0.229 (T/D) -0.852 (d/D) -0.861 (t/T) 0.543 sin 0.210 T-1 T-1 T-1 T-1 T-1 T-1 T-1 T		Tof T,K = 4.491 (7/6) -0.123 (d) -0.396 (t/) 0.672 (9,+92) 0.153 sin 2.2674	Τ-
Kcharl = 1.40(T/o)^-0.38 (d/o)^0.06 (t/T)^0.94 sin^0.94 Kstrauch = 2.877 (d/o)^-0.35 (t/T)^0.35 sin^0.5 to T-1 T,YCL = 0.507 (T/o)^-1.014 (d/o)^-0.857 (t/T)^0.889 sin^1.557 to T-1 0.3 < d/o < 0.55 T,YCL = 0.229 (T/o)^-1.014 (d/o)^-0.419 (t/T)^0.889 sin^1.557 to T-1 0.55 & d/o & 0.75 TRANSVERSE T,YStrauch = 0.243 (T/o)^-0.852 (d/o)^-0.801 (t/T)^0.543 sin^2.035 to T-1 88NDING 0.3 < d/o < 0.55 T,Y Branch = 0.441 (T/o)^-0.852 (d/o)^-0.281 (t/T)^0.543 sin^2.035 to T-1	2	T,Ychord = 0.463 (T/6) -0.6 (d/6) -0.04 (+/) 0.86 sin 0.57 &	7-
Tycher = 0.507 (7/0) -0.35 (t/T) 0.35 sin 0.5 A Tycher = 0.507 (7/0) -0.014 (d/0) 0.787 (t/T) 0.889 sin 1.557 A Tycher = 0.229 (T/0) -1.014 (d/0) -0.619 (t/T) 0.889 sin 1.557 A Trunsverse Trunsverse Tybranch = 0.843 (T/0) -0.852 (d/0) 0.801 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A Tybranch = 0.441 (T/0) -0.852 (d/0) -0.281 (t/T) 0.543 sin 2.035 A	N-PLANE BENDING		T-
T,YCL = 0.507 (T/p) -1.014 (d/p) 0.787 (t/T) 0.889 sin 1.557 f 0.3 < d/p < 0.55 T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.619 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -1.014 (d/p) -0.819 (t/T) 0.889 sin 1.557 f T,YCL = 0.229 (T/p) -0.852 (d/p) -0.819 (t/T) 0.543 sin 2.033 f T,YCL = 0.229 (T/p) -0.852 (d/p) -0.281 (t/T) 0.543 sin 2.033 f T,YCL = 0.241 (T/p) -0.852 (d/p) -0.281 (t/T) 0.543 sin 2.033 f T,YCL = 0.241 (T/p) -0.852 (d/p) -0.281 (t/T) 0.543 sin 2.033 f	RA	Kenned = 1.40 (T/0)-0.38 (d/0)0.06 (t/) 0.94 sin 0.90	Τ-
0.3 < d/D < 0.55 T.YCHOM = 0.229 (T/D)-1.014 (d/D)-0.619 (t/T) 2889 Sin 1.557 A T-1 O.55 & d/D & 0.75 T.YBranch = 0.843 (T/D)-0.852 (d/D) 0.801 (t/T) 0.543 sin 2.035 A T.YBranch = 0.441 (T/D)-0.852 (d/D) 0.801 (t/T) 0.543 sin 2.035 A T.YBranch = 0.441 (T/D)-0.852 (d/D)-0.281 (t/T) 0.543 sin 2033 A T.YBranch = 0.441 (T/D)-0.852 (d/D)-0.281 (t/T) 0.543 sin 2033 A T-1	M.PLANE BENDING	Kananch = 2.87 (d/b)-0.35 (t/T) 0.35 sin 0.50	T-
Trunsverse Trunsv			·T-
TRANSVERSE TYBranch = 0.243(T/D)-0.852(d/D)0.801(t/T)0.543 sin 2.0350 T-1 BENDING 0.3 < d/D < 0.55 TYBranch = 0.441 (T/D)-0.852 (d/D)-0.281(t/T)0.543 sin 2.0330 T-1		T,YCLON = 0.229 (T/D)-1.014 (d/D)-0.619 (t/T) 0889 sin 1.557 &	Τ-
T, Y Brand = 0.441 (T/0) -0.852 (4/0) -0.281 (+/T) 0.543 sin 20330 T-1		T. Yoranch = 0.843 (T/D) -0.852 (d/D) 0.801 (t/T) 0.543 sin 2.033 A	T-1
		T, Y Brand = 0.441 (T/0) -0.852 (d/0) -0.281 (=/T) 0.543 sin 2033 0	T-1
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		SCF in ch	hord		SCFinb	race		4	
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HE I I I I I	▲ Smedley	5.87	6.66	4.98	4.70	5.20	4.14	-	
E (15%)				In-pla	ne bending			40.00	
	Gibstein	2.03	2.03	1.85	2.03	2.02	1.83	-	
	Kuang	1.59	1.55	1.53	2.59	2.06	1.78		
	& Smedley	2.14	2.23	2.04	2.35	2.40	2.29		
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	Gibstein	2.91		. 4.32	2.78	4.64	4.93		
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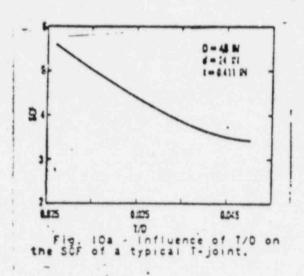




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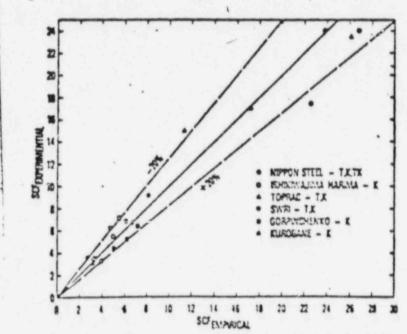
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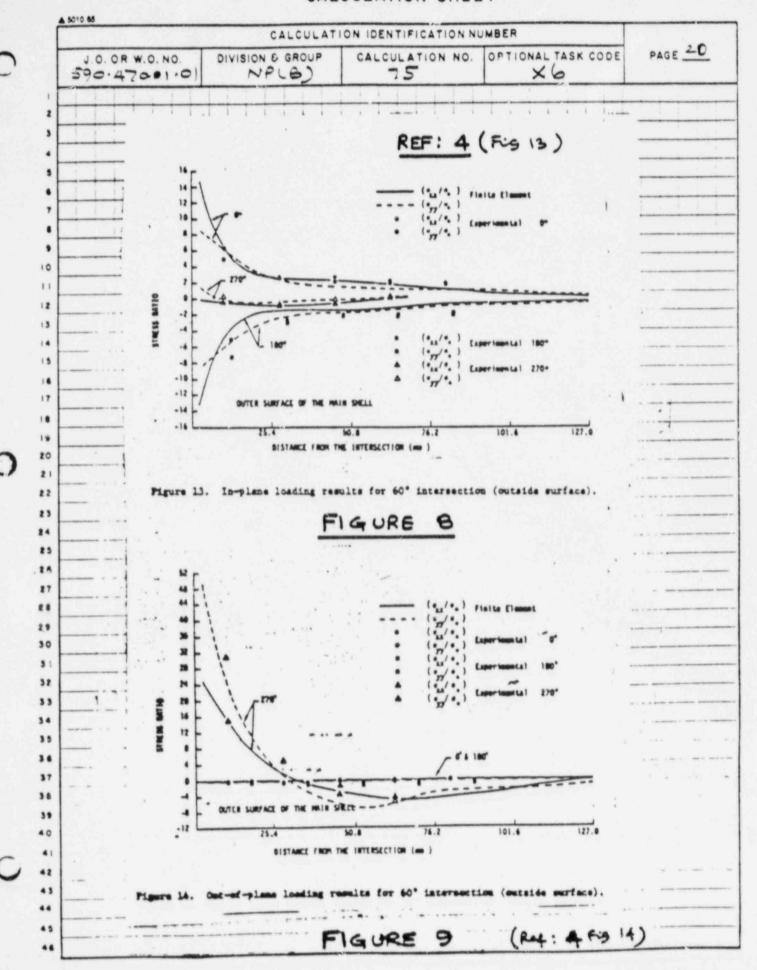
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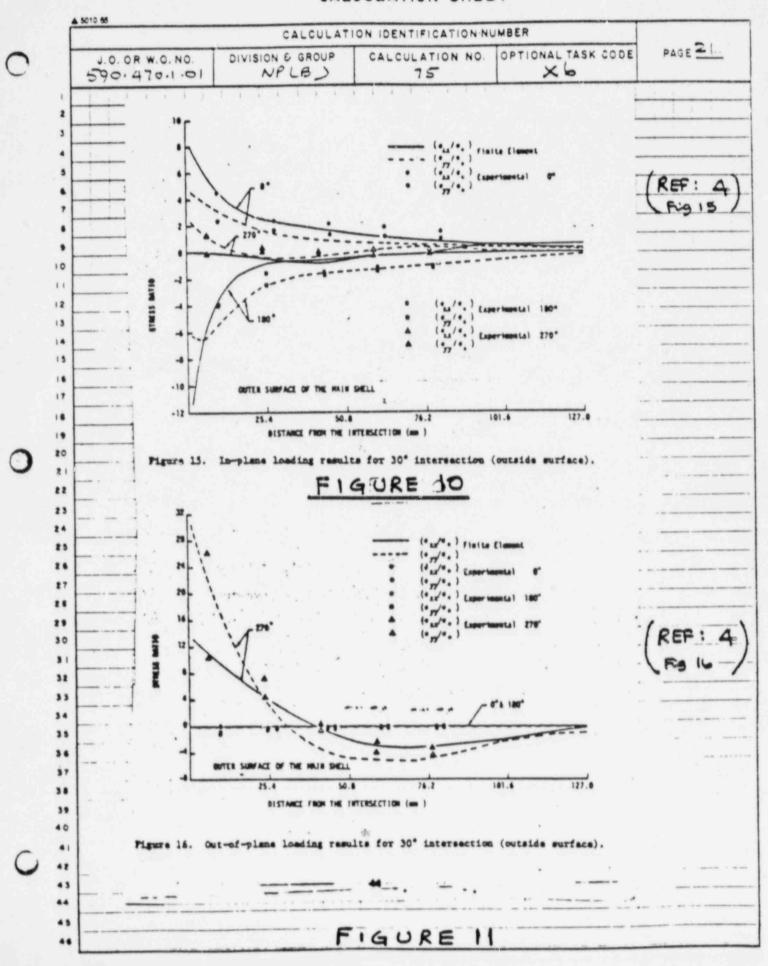
FIGURE G

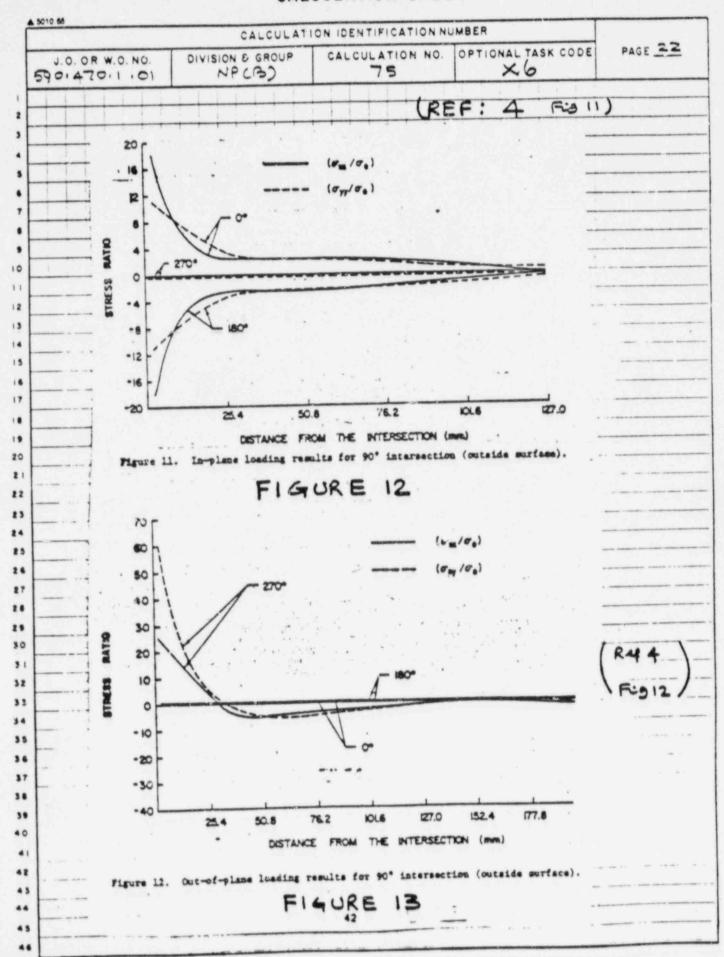


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Fig. 11 - Comparison of experimental and empirateal SCFs for various joints.







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