

GA Technologies Inc.

GA 1488 (REV. 10/82)

## **ISSUE SUMMARY**

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1. SUMMARY

The minimum numbers of tendons required for safely supporting the core cavity pressure of 845 psig (Reference Pressure, RP) and 1268 psig (1.5 RP) without breaching the liner of the Fort St. Vrain PCRV have been determined for: 1) circumferential tendons in the PCRV wall and 2) crosshead and circumferential tendons in the heads of the PCRV. Hand calculations based on the concept of ultimate load analysis are used. No calculation is performed for the vertical tendons in this study. Per Ref. 1, the PCRV can resist up to 1515 psig cavity pressure with only the rebars acting, i.e., without reliance on any vertical prestress.

The results are given in Table 1 and Figs. 1 and 2. These results indicate that the core cavity pressure of 1.0 RP can be safely resisted with considerably less number of tendons than is actually provided. With 1.5 RP, the number of head tendons required is still less than that actually provided. The difference, however, is small if no vertical tendons exist as assumed in the analysis. Existence of vertical tendons will require less number of head tendons to resist 1.0 and 1.5 RP.

The procedure used in the study is described in the following sections. Detailed calculations are given in the appendices.

2. MATERIAL PROPERTIES

Material properties used in this analysis are (Ref. 1):

Concrete:

Compressive strength  $f'_c = 6000 \text{ psi}$

Liner:

Material = SA 537, Gr. B

Yield strength  $f_{sy} = 60,000 \text{ psi}$  (at 0.2% offset)

Tensile strength  $f'_s = 80,000 \text{ psi}$  (Ref. 2)

Failure strain  $\epsilon' = 18\%$  (Ref. 2)

Modulus of elasticity  $E = 29 \times 10^6 \text{ psi}$

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Tendon Wires:

Tensile strength  $f'_s$  = 240,000 psi

Yield strength  $f_{sy}$  = 204,000 psi (at 1% strain)

Failure strain  $\epsilon'$  = 4%

Modulus of elasticity  $E$  =  $27 \times 10^6$  psi

Rebars:

Material = A432

Tensile strength  $f'_s$  = 90,000 psi

Yield strength  $f_{sy}$  = 60,000 psi

Failure strain  $\epsilon'$  = 7%

Modulus of elasticity  $E$  =  $29 \times 10^6$  psi

3. DEFINITION OF NUMBERS OF TENDONS

The Fort St. Vrain PCRV has, in addition to 90 vertical tendons with 169 1/4-in. diameter wires each, 210 circumferential tendons with 152 1/4-in. diameter wires each, 100 circumferential tendons with 169 1/4-in. diameter wires each, and 48 crosshead tendons with 169 1/4-in. diameter wires each.

All 210 152-wire circumferential tendons are in the barrel section. Of the 100 169-wire circumferential tendons, 34 are in the top head (the top 15'-6" section), 34 in the bottom head (the bottom 15'-6" section), and 16 each in the barrel sections adjacent to the top and bottom heads. Each head has 24 cross-head tendons.

All circumferential tendons are 180° tendons rather than full circle tendons (see Fig. E.15-2, Ref. 1). Because of the arrangement of these tendons, of the 18 circumferential (180°) tendons in a typical five-foot high wall section, a minimum of 12 pass any cross section. Hence 18 actual circumferential tendons provide 12 "effective" circumferential tendons. Similarly in the top or bottom head, 34 actual circumferential tendons provide 22 effective circumferential tendons.

The following definitions are used in this report:

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$N_b$  = Number of effective circumferential tendons in a 5-foot high wall section (12 in existence).

$N_c$  = Number of effective circumferential tendons in the top or bottom head (22 in existence).

$N_x$  = Number of crosshead tendons in the top or bottom head (24 in existence).

It is assumed that there is no broken wire in any tendon and that the required tendons in each group are uniformly distributed.

#### 4. CIRCUMFERENTIAL TENDONS IN PCRV WALL

For the determination of the required number of circumferential tendons in the PCRV wall a typical five-foot high wall section was considered. It is assumed that ultimate conditions are reached at 1.0 RP or 1.5 RP for the purpose of this analysis. The core cavity liner is anchored to the concrete by means of studs welded to the liner and embedded in the concrete. The stud spacings are 7-1/2 in. in both circumferential and axial directions. It is assumed that, at ultimate, radial concrete cracks would develop at stud anchor locations and that resistance to the core cavity pressure is provided by the steel elements acting as multiple structural rings. The steel elements include the liner, and circumferential tendons and rebars at various radial locations. With the liner and rebar cross-sectional areas known, the number of tendons required to provide a total pressure resistance capacity for the core cavity pressure of 1.0 RP or 1.5 RP, and meeting the selected limit criteria can be determined from equilibrium and strain compatibility.

The tendon prestress loss at end of life is assumed to be 13.5% (Ref. 1), and the friction loss is assumed to be 11.5% (Ref. 3) in these calculations.

Two limit criteria are used in this case:

1) Liner stress =  $0.9 f_{sy}$ ,

Tendon stresses  $\leq f'_s$ , and

Rebar stresses  $\leq f'_s$

2) Maximum tendon stress =  $f'_s$ ,

Liner stress  $< f'_s$ , and

Rebar stresses  $\leq f'_s$

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Obviously the first criterion is the more stringent and results in a larger number of tendons being required. The required numbers of circumferential tendons in the PCRV wall for the above two limit conditions are shown in Table 1.

#### 5. TENDONS IN PCRV TOP AND BOTTOM HEADS

The required number of crosshead and circumferential tendons in the PCRV top and bottom heads to safely support 1.0 RP and 1.5 RP without breaching of the liner boundary are determined by ultimate load analysis of the bottom head.

Four quasi-analytical solutions were originally used in assessing the ultimate capacity of the Fort St. Vrain PCRV heads (Ref. 1). These are: 1) bottom head yield line failure analysis, 2) bottom head punching shear failure analysis, 3) bottom head concrete ligament compressive failure analysis, and 4) top head analysis by grid system simulation. In the case of 34 circumferential and 24 crosshead tendons in each head and 90 vertical tendons, the yield line analysis provided the lowest estimate of the ultimate pressure capacity, while the concrete ligament compressive failure analysis provided the highest, about three times as high as the lowest estimate. The top head grid analysis requires use of a computer program.

Based on the above observations it was decided to use the yield line failure analysis method for the ultimate load analysis in the current study, and to check the results using the punching stress failure analysis.

The assumptions and detailed procedure used in the bottom head yield line analysis follow those used in Ref. 1. Based on an assumed number of crosshead tendons the resultant pressure which must be resisted by the bottom head (cavity pressure reduced by the cavity pressure equivalent of crosshead tendons, Ref. 1) is first calculated. By assuming formation of a plastic hinge at the head-to-wall junction (signified by 0.003 in/in maximum concrete strain and/or yielding of liner and majority of rebars in tension), and a yield line pattern (generally radial along concrete ligaments) the unit yield line moment required to prevent this particular yield line mode of failure under the given cavity pressure (1.0 RP or 1.5 RP) can be determined. The number of circumferential tendons required to provide an ultimate moment capacity along the yield line which is larger than the required unit yield line moment is then established. The ultimate moment capacity of the bottom head is defined by the following stress limits (Ref. 1):

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Maximum rebar stress  $\leq 0.9 f_{sy}$ ,

Maximum tendon stress  $\leq 0.9 f_{sy}$ , and

Maximum compressive concrete stress  $\leq 0.85 f'_c$

The tendon prestress losses at end of life are assumed to be 12% for both crosshead and circumferential tendons (Ref. 1) and the friction losses are assumed to be 10% and 11.5%, respectively for these two types of tendons (Ref. 3). The yield line failure analysis in the current study is based on the assumption that no vertical tendons exist.

The required number of tendons based on the results of the yield line failure analysis are shown graphically in Figs. 1 and 2.

For the punching shear failure analysis of the bottom head the failure plane is assumed to be the one formed by the concrete ligaments connecting the steam generator penetrations (Ref. 1). Reference 4 provides an equation to estimate the ultimate shearing strength of PCRV heads as a function of span/depth ratio and radial prestress. Based on this equation and the number of head tendons required as determined by the yield line failure analysis, it is found that the punching shear stress is not critical for either the 1.0 RP or the 1.5 RP cases.

## 6. CONCLUSIONS

The required numbers of circumferential tendons in the Fort St. Vrain PCRV wall to safely support the cavity pressure of 1.0 RP and 1.5 RP are given in Table 1. The corresponding required numbers of crosshead and circumferential tendons in either top or bottom head of the PCRV, derived under a conservative assumption of no vertical prestress, are given in Figs. 1 and 2.

From Fig. 2, it appears that under 1.5 RP the permissible reduction in the numbers of crosshead and circumferential tendons in the heads is small if no vertical prestressing tendons exist.

## 7. REFERENCES

1. "Fort St. Vrain Nuclear Generating Station. Updated Final Safety Analysis Report."
2. ASTM, "Specification for Carbon-Manganese-Silicon Steel Plates, Heat Treated for Pressure Vessels. SA-537."

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3. Lee, T. T. and Cheung, K. C., "FSV - PCRV Tendon Evaluation," GA Document 907441/A, April 30, 1984.
4. Garas, F. K. and Trowsdale, D. R., "Overload Behavior and Shear Failure Mechanisms of Model No. 2 of the Bottom Head of the Fort St. Vrain Prestressed Concrete Reactor Vessel," Report 14H/69/1411, Taylor Woodrow Construction, Ltd., September 1969.
5. Bresler, B., "Reinforced Concrete Engineering," Vol. 1 Materials Structural Elements, Safety, John Wiley & Sons, New York, 1974.
6. "PCRV Bottom Head, Reinforcing Plan, Sheet 1," Drawing 3614, B-36/J, Sargent & Lundy, March 1969.
7. "PCRV Bottom Head, Reinforcing Schedule and Details, Sheet 1," Drawing 3614, B-37/E, Sargent & Lundy, October 1969.
8. "PCRV Bottom Head, Reinforcing Plan, Sheet 2," Drawing 3614, B-38/K, Sargent & Lundy, April 1969.
9. "PCRV Bottom Head, Reinforcing Schedule & Details, Sheet 2," Drawing 3614, B-39/E, Sargent & Lundy, December 1968.
10. "PCRV Bottom Head, Reinforcing Schedule & Details, Sheet 3," Drawing 3614, B-40/D, Sargent & Lundy.
11. "PCRV Bottom Half Vertical Section," Drawing 3614, B-35/S, Sargent & Lundy, December 1969.
12. "PCRV Bottom Head, Tendon Tubes Details," Drawings 3614, B-21/E and B-22/D, Sargent & Lundy, October 1969.

NOTE: References 5 through 12 are cited in the appendices.

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TABLE 1

## MINIMUM NUMBERS OF CIRCUMFERENTIAL TENDONS IN WALL SECTION

Criterion	Number of Tendons Required ( $N_b$ ) <sup>(1)</sup>		Percentage of Tendons Required <sup>(2)</sup>	
	1.0 RP <sup>(3)</sup>	1.5 RP	1.0 RP	1.5 RP
Liner Stress = $0.9 f_{sy}$	5	9	42%	75%
Max. Tendon Stress = $f'_{s}$ <sup>(4)</sup>	3	5	25%	42%

(1) Number of effective tendons required per 5-foot high section. See the text for definition of  $N_b$ .

(2) Percentage of tendons currently provided in any region of the PCRV wall. It is assumed that the required tendons are located uniformly in the region under consideration.

(3) 1.0 RP = 845 psig.

(4) The liner strain is 0.046 in./in. when the maximum tendon stress is  $f'_{s}$ .

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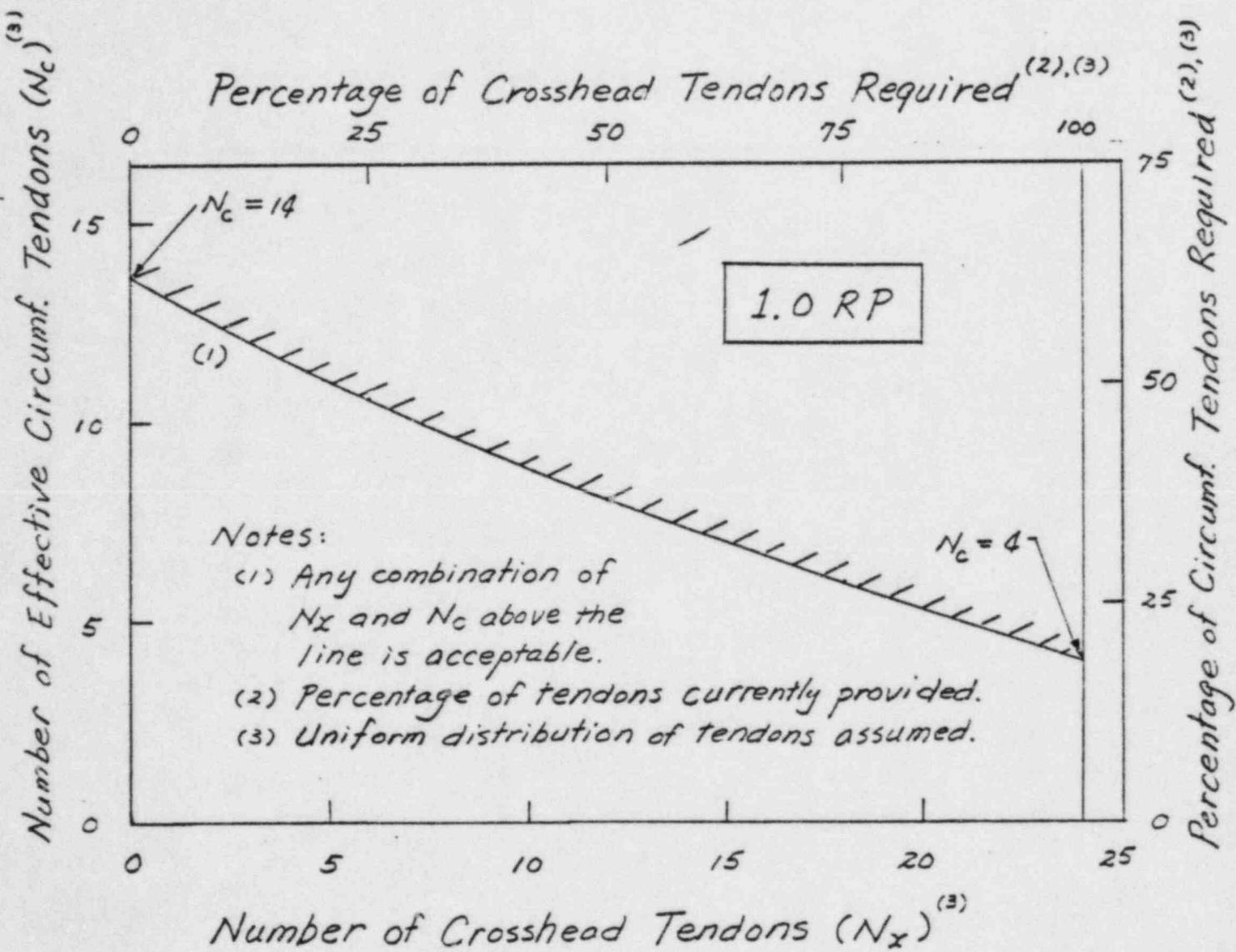


Figure 1. Number of Head Tendons Required in Each Head to Support 1.0 RP

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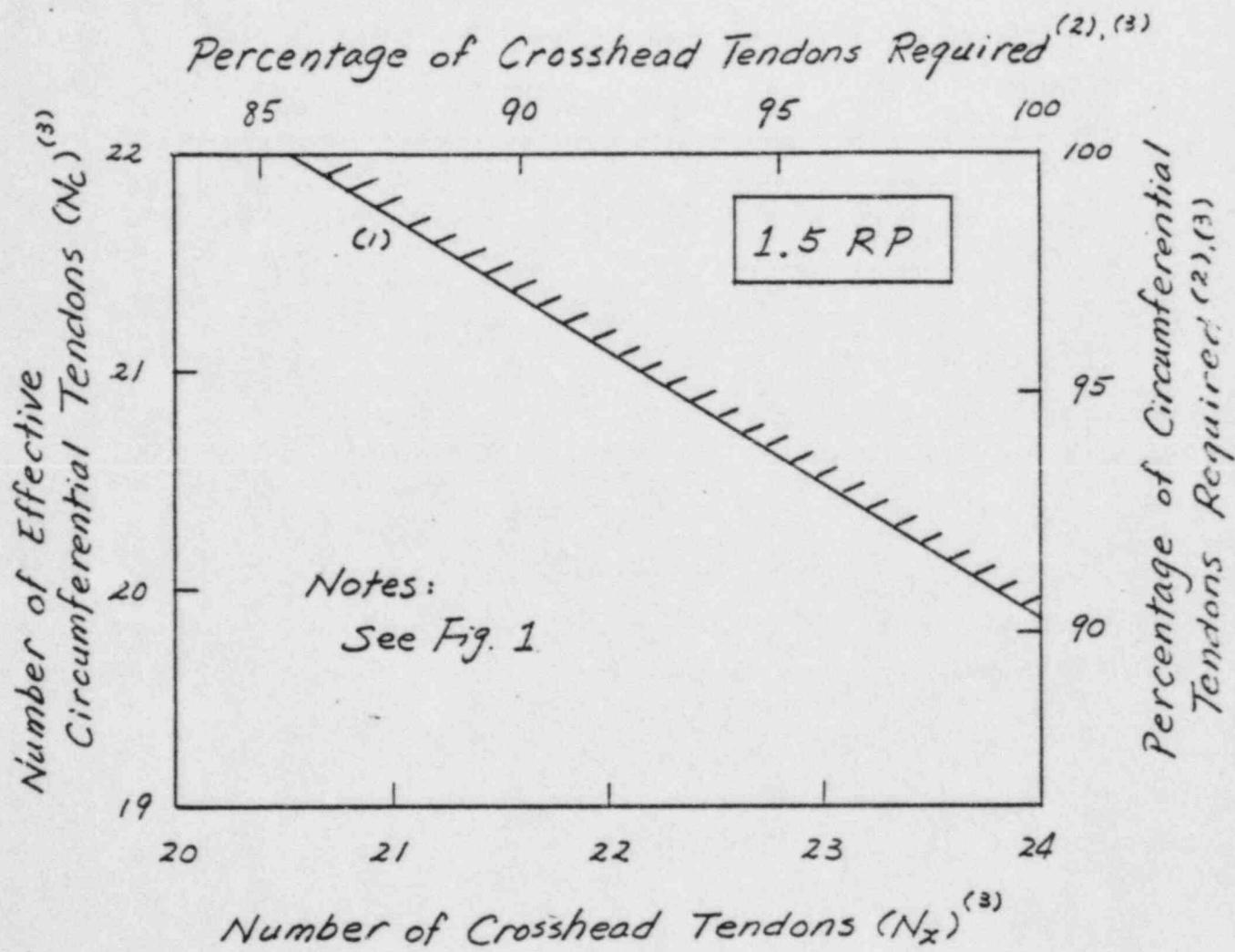


Figure 2. Number of Head Tendons Required in Each Head to Support 1.5 RP

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APPENDIX A

NUMBER OF CIRCUMFERENTIAL TENDONS IN THE PCRV WALL

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### A.1 Cases Considered

$$1) \text{ Core cavity pressure} = \text{Reference Pressure} \\ = 1.0 R P = 845 \text{ psig}$$

$$2) \text{ Core cavity pressure} = 1.5 RP = 1268 \text{ psig.}$$

## A.2 Location of Steel Components

For a 5'-0" barrel section.

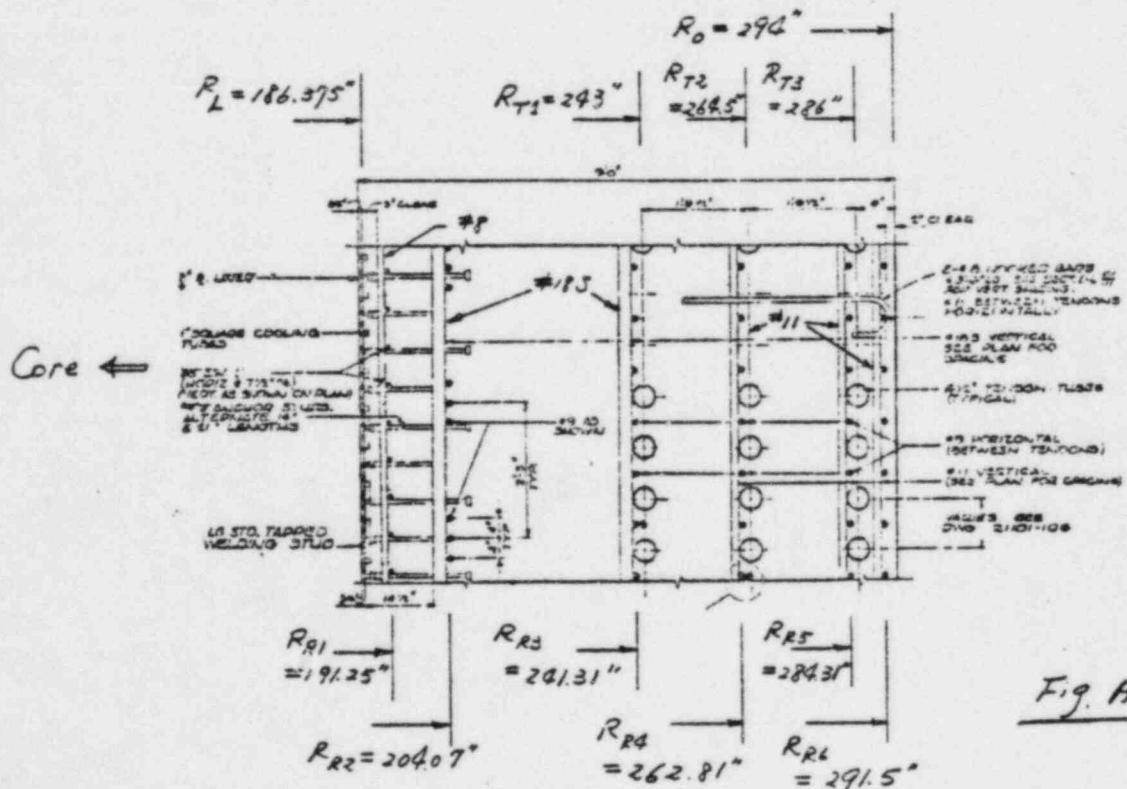


Fig. A-1

(Ref. 1. FSAR FIG. E.15-6 )

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## 3 Rebar Dimensions:

4 Table A-1.

Bar Size	Diam. (in.)	Area (in. <sup>2</sup> )
# 8	1.000	0.79
# 9	1.128	1.00
# 11	1.410	—
# 18 S	2.257	—

## 19 Rebar Groups: Table A-2

Group	R <sub>R</sub> (in.)	Rebars	Number per 5 ft.	Total Area (in. <sup>2</sup> )
R1	191.25	#8 @ 7.5"	8	6.32
R2	204.07	3 - #9 @ 27"	6	6.00
R3	241.31	#9 @ 10"	6	6.00
R4	262.81	#9 @ 10"	6	6.00
R5	284.31	#9 @ 10"	6	6.00
R6	291.50	#8 @ 10"	6	2.74

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Liner:  $\frac{3}{4}$  in. thick at  $R_L = 186.375$  in.

Tendons: 152 -  $\frac{1}{4}$  in. diam. wires  $\rightarrow$  Area = 7.463  
169 - " " " "  $\rightarrow$  Area = 8.298

- 3 groups of 6 tendons each (total of 18)

for each 5 ft. section, located at  $R_{T1}$ ,  $R_{T2}$ , &  $R_{T3}$ .

- Minimum of 12 tendons pass any 5 ft. section.

### A.3 Material Properties

Rebars: A432. (A305-S6T. Deformation)

$f_{sy}$  = Minimum guaranteed yield strength

(at 0.5% total extension)

= 60 ksi. (FSAR, App. E. Section E.1)

$f_s'$  = Minimum guaranteed tensile strength

= 90 ksi (FSAR, Section E.1)

$\epsilon'$  = Minimum guaranteed strain at failure

= 7% in  $\delta$  in. (FSAR, Section E.1)

$E$  = Modulus of elasticity =  $29.0 \times 10^3$  ksi.  
(FSAR, Sec. E.10.3)

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Liner: SA 537, Grade B (Quenched & tempered)  
(Ref. 1)

$f_{sy} = 60$  ksi (at 0.2% offset) (FSAR, E.1)  
(FSAR, E.102)

$$f_s' = 80 \text{ ksi} \quad \left\{ \text{(ASTM Spec. SA 537, Ref. 2)} \right.$$

$$\epsilon' = 18\%$$

$$E = 29.0 \times 10^3 \text{ ksi} \quad (\text{FSAR Sec. E.10.3})$$

### Tendons:

Tendon wires:  $f_s' = 240 \text{ ksi}$

$$f_{sy} = 204 \text{ ksi at } 1\% \text{ strain}$$

$$\epsilon' = 49.$$

See Fig. 4 (From FSAR, Fig. 5.6-1)

$$E = 27 \times 10^3 \text{ ksi} \quad (\text{Table 5.6-5})$$

PSAR

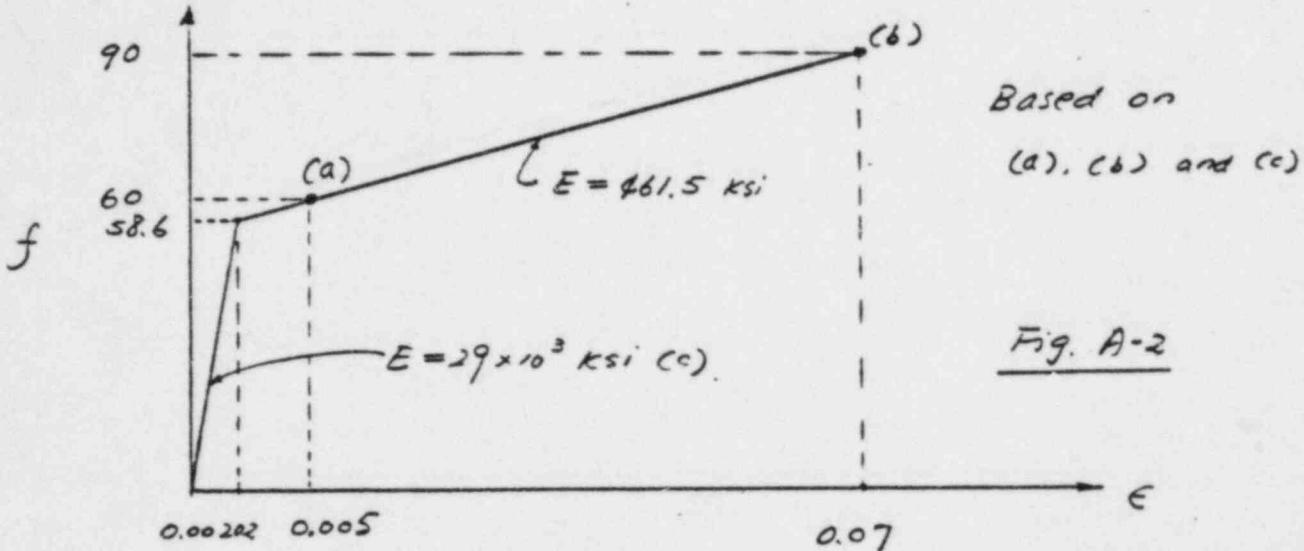
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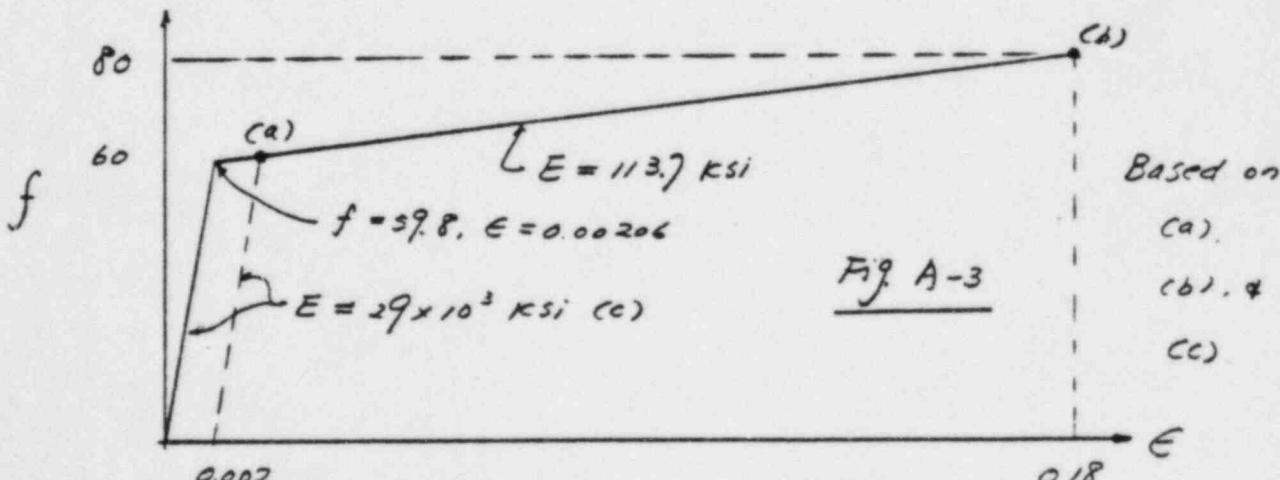
1  
2 Stress - Strain Curves:  
3

4 Rebars:



use  $\left\{ \begin{array}{l} f = 90 - (0.07 - \epsilon) \times 461.5, \quad \epsilon \geq 0.00202 \\ f = 29.0 \times 10^3 \epsilon, \quad \epsilon \leq 0.00202 \end{array} \right.$

Liner:



use  $\left\{ \begin{array}{l} f = 80 - (0.18 - \epsilon) \times 113.7, \quad \epsilon \geq 0.00206 \\ f = 29.0 \times 10^3 \epsilon, \quad \epsilon \leq 0.00206 \end{array} \right.$

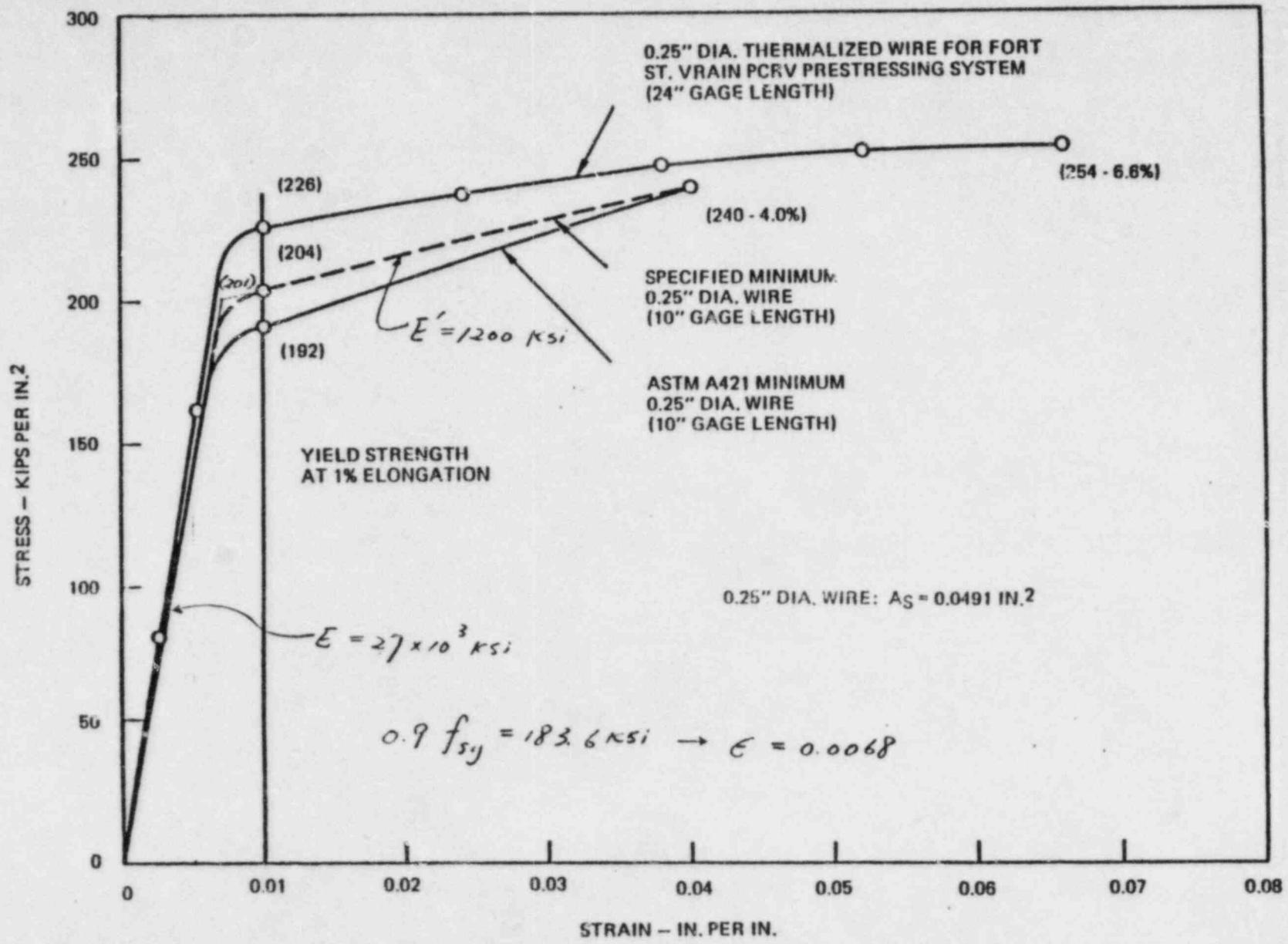


Figure A-4 Typical Stress-Strain Curve of Wire

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A.4 Assumptions

- (1) Based on the discussion on the crack propagation given in FSAR, Section E.8.2.3.1, it is assumed that radial cracks develop at the locations of stud anchors. Hence concrete does not participate in resisting the core cavity pressure. It does transmit the pressure radially.
- (2) Each steel component (liner, rebar and tendon) acts as a ring and contributes in resisting the core cavity pressure.
- (3) On account of the extensiveness of the concrete crack the bond between rebars and concrete is essentially non-existing and the strain in a given rebar is essentially

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constant (Ref. 5, p 175).

(4) The prestress loss at end of life is

13.5% (FSAR, Table 5.6-4). The friction loss  
is 11.5% (Ref. 3)

(5) Radial shortening of concrete is negligible.

Hence the radial displacements of all steel  
components are the same.

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A.5 Compatibility

Assuming that each steel component displaces  
 radially the same amount, the strains in  
 various components are related as follows:

$$\epsilon_L R_L = \epsilon_{Ri} R_{Ri} = (\Delta \epsilon_{Ti}) R_{Ti}$$

where  $\epsilon$  = strain

$R$  = radial distance from the center  
 of core cavity

$\Delta \epsilon_{Ti}$  = additional tendon strain for Group i  
 tendon  
 = total tendon strain - strain due  
 to prestress.

The subscripts are:

$L$  = liner

$R$  = a rebar group

$T$  = a tendon group

$i$  = group number.

The strain due to prestress  $\cong 0.00476$  in./in. (Fig A-4)  
 based on  $f = 0.7 f'_s \times (1 - 0.135) \times 0.885 = 128.6$  ksi

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A.6 Contribution to Pressure Capacity

Let  $f$  = stress (psi)

$A_s$  = total steel area ( $in^2$ )

$p'$  = pressure capacity if the pressure  
is applied to the steel component.  
(psi)

$p$  = contribution to the core cavity  
pressure capacity (psi)

Liner:

$$\frac{P'_L R_L \cdot 60}{A_{s,L}} = f_L \rightarrow P'_L = \frac{3/4 \times 60}{186.375 \times 60} f_L = 4.024 \times 10^{-3} f_L$$

$$P_L = P'_L = 4.024 \times 10^{-3} f_L$$

Rebar Groups:

$$\frac{P'_{Ri} R_{Ri} \cdot 60}{A_{s,Ri}} = f_{Ri} \rightarrow P'_{Ri} = \frac{A_{s,Ri}}{60 \cdot R_{Ri}} f_{Ri}$$

$$P_{Ri} = P'_{Ri} \frac{R_{Ri}}{R_L} = \frac{A_{s,Ri}}{60 R_L} f_{Ri}$$

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For  $R_1$ :  $P_{R1} = \frac{6.32}{60 \cdot 186.375} f_{R1} = 0.565 \times 10^{-3} f_{R1}$

For  $R_2 \sim R_5$ :

$$P_{Ri} = \frac{6.00}{6.32} \cdot 0.565 \times 10^{-3} f_{Ri} = 0.536 \times 10^{-3} f_{Ri}$$

$$\text{For } R_6: P_{R6} = \frac{4.74}{6.32} \cdot 0.565 \times 10^{-3} f_{R6} = 0.424 \times 10^{-3} f_{R6}$$

Tendon Groups:

$$\frac{P'_{Ti} R_{Ti} 60}{A_{s,Ti}} = f_{Ti} \rightarrow P'_{Ti} = \frac{A_{s,Ti}}{60 R_{Ti}} f_{Ti}$$

$$P_{Ti} = P'_{Ti} \frac{R_{Ti}}{R_L} = \frac{A_{s,Ti}}{60 R_L} f_{Ti}$$

$$= \frac{N' \cdot 7.463}{60 \cdot 186.375} f_{Ti} = 0.667 \times 10^{-3} N'_i f_{Ti} \quad (\text{152-wire})$$

$$= \frac{N' \cdot 8.298}{60 \cdot 186.375} f_{Ti} = 0.742 \times 10^{-3} N'_i f_{Ti} \quad (\text{169-wire})$$

where  $N'_i$  = minimum number of tendons  
passing any 5 ft section

(for each group, maximum is 4)

## CALCULATION SHEET

## CALCULATIONS FOR

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A.7 Calculation of Minimum Tendons Required

- (1) Select a limit criterion in terms of  $\epsilon$ .
- (2) Calculate strains for all steel components using the compatibility relations. Check  $\epsilon \leq \epsilon'$ .
- (3) Calculate  $f_L$ ,  $f_{R,i}$  and  $f_{T,i}$  (Figs. A-2 ~ A-4)
- (4) Calculate  $P_L$ ,  $P_{R,i}$  and  $P_{T,i}$
- (5) Calculate the total pressure capacity to be contributed by tendons:

$$\left( \sum_{i=1}^3 P_{T,i} \right)_{\text{req'd}} = 845 - (P_L + \sum_{i=1}^6 P_{R,i})$$

(cor 1268)

- (6) Calculate the total number of tendons required, using the middle row as an average:

$$N_b = \sum_{i=1}^3 N'_i = \frac{\left( \sum P_{T,i} \right)_{\text{req'd}}}{0.667 f_{T2}}$$

(152-wire) (cor 0.742) ~ (169-wire tendons)

$f_{T2}$  in kpsi.

$$\text{Or } N = \frac{18}{12} \sum_{i=1}^3 N'_i$$

where  $N$  is the actual total number of tendons required to provide  $\sum_{i=1}^3 N'_i$  effective number of tendons in any 5 ft section.

## CALCULATION SHEET

CALCULATIONS FOR

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1RP Case

Table A-3.

Criterion: Liner Stress at 0.9  $f_{sy}$  = 54 ksi

Component	R (in.)	$\frac{R_L}{R}$	$\Delta \epsilon_{Ti}$	$\epsilon$	f (ksi)	P (psi)
Liner	186.38	1		0.00186	54.0	217.3
R1	191.25	0.975		0.00181	52.5	29.7
R2	207.07	0.913		0.00170	49.3	26.4
R3	241.31	0.772		0.00144	41.8	23.6
R4	262.81	0.709		0.00132	38.3	20.5
R5	284.21	0.656		0.00122	35.4	19.0
R6	291.50	0.639		0.00119	34.5	14.6
T1	243.00	0.767	0.00123	0.00619		
T2	264.50	0.705	0.00131	0.00607	Average 164	109 each tension
T3	286.00	0.652	0.00121	0.00597		

$$845 - (P_s + \sum P_{Ri}) = 845 - (351.1) = 493.9 \text{ psi}$$

$$\sum_{i=1}^3 N_i = \frac{493.9}{109} = 4.53 \quad \rightarrow N = 6.80$$

$$\text{For } N = 7 \quad P = 351 + 109 \times 7 \times \frac{2}{3} = 860 \text{ psi}$$

## CALCULATION SHEET

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1 RP Case

Table A-4

Criteria 2:

Tendons at R = 243 in. have  $\epsilon = \epsilon'$ 

Component	R (in.)	$\frac{R_{Ti}}{R}$	$\Delta\epsilon_{Ti}$	$\epsilon$	f (ksi)	P (psi)
Liner	186.38	1.30		0.0458	64.7	260.5
R1	191.25	1.27		0.0447	78.3	222
R2	209.07	1.19		0.0419	77.0	41.3
R3	241.31	1.01		0.0356	74.1	39.7
R4	262.81	0.925		0.0326	72.7	39.0
R5	284.31	0.855		0.0301	71.6	38.4
R6	291.50	0.834		0.0294	71.3	30.2
T1	243.00	1	0.0352	0.04	240	160.0 each
T2	264.50	0.919	0.0324	0.0372	236.6	157.8 each tendon
T3	286.00	0.850	0.0300	0.0347	233.6	155.8 each

$$\delta 45 - (P_L + \sum P_{Ti}) = \delta 45 - (493) = 352$$

$$\sum_{i=1}^3 N_i' = 352 / 157.8 = 2.23$$

$$\rightarrow N = 3.35$$

$$\text{For } N=4, P = 4P_3 + 4 \times 157.8 \times \frac{2}{3} = 916 \text{ psi}$$

## CALCULATION SHEET

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1 R P Case Table TA-5

Criterion 3:

Rebars at  $R = 191.25$  in. have  $\epsilon = \epsilon'$ 

Component	$R$ (in.)	$\frac{R_{Ri}}{R}$	$\Delta\epsilon_{Ti}$	$\epsilon$	$f$ (ksi)	$P$ (psi)
Liner	186.38	1.03		0.072	67.7	272.5
$R_1$	191.25	1		0.07	90.0	50.9
$R_2$	209.07	0.937		0.0656	87.8	47.2
$R_3$	241.31	0.793		0.0555	83.3	29.6
$R_4$	262.81	0.728		0.0509	81.2	23.5
$R_5$	284.31	0.673		0.0471	79.4	22.6
$R_6$	291.50	0.656		0.0459	78.9	23.4
$T_1$	243.00	0.787	0.0551	0.0599		0
$T_2$	264.50	0.723	0.0506	0.0554	$\} > 6\% = \epsilon'$	0 each tendon
$T_3$	286.00	0.669	0.0468	0.0516		
					Failed.	0

$$845 - (P_L + \sum P_{Ri}) = 845 - (535) = 310$$

$$\sum_{i=1}^6 N_i' = N.A.$$

$$\rightarrow N = N.A$$

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Tables A-3 through A-5 are for tendons with 152 wires. For tendons with 169 wires, still for IRP, using the data in Tables A-3 and A-4:

Criterion 1.  $f_t = 167 \text{ ksi} \rightarrow \frac{T_{Ti}}{N_i} = 121.7 \text{ psi}$

$$\sum N_i' = \frac{493.9}{121.7} = 4.06 \text{ vs } 4.53 \text{ for } 152\text{-wire tendons}$$

Criterion 2  $f_t = 236.6 \rightarrow \frac{T_{Ti}}{N_i} = 175.6$

$$\sum N_i' = \frac{352}{175.6} = 2.00 \text{ vs } 2.23 \text{ for } 152\text{-wire tendons}$$

→ Use the same number of tendons  
as in the 152-wire tendon case.

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1.5 RP Case

Data in Tables A-3 and A-5 still apply.

The required pressure capacity is 1268 psig instead of 845 psig.

Based on Table A-3.

$$P_e + \sum P_{ei} = 351.1 \text{ psi}$$

Tendons must supply  $1268 - 351.1 = 916.9$

$$\sum_{i=1}^3 N_i' = \frac{916.9}{109} = 8.41 \rightarrow N = 12.6$$

$$\text{Use } N = 13 \rightarrow P = 351.1 + 109 \times 13 \times \frac{2}{3} = 1293 \text{ psi}$$

Criterion 2: Tendons at R = 243 in. have  $\epsilon = \epsilon'$

$$\sum_{i=1}^3 N_i' = \frac{1268 - 1293}{157.8} = 4.91 \rightarrow N = 7.37$$

$$\text{Use } N = 8 \rightarrow P = 1293 + 8 \times 157.8 \times \frac{2}{3} = 1335 \text{ psi}$$

Criterion 3: Rebars at R = 191.25 in have  $\epsilon = \epsilon'$

Tendon strain > 4%

$P = 535 \text{ psi}$

N.G.

## CALCULATION SHEET

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A.8 Summary of Calculated Results

Table A-6

Minimum Number of Circumferential Tendons  
Required per 5 ft. High Wall Section

Limit Criteria	Required 1.0 RP		Required 1.5 RP		Number currently Provided	
	Actual	Effective	Actual	Effective	Actual	Effective
Liner Stress $= 0.9 f_{sy}$	7	5	13	9	18	12
Max. Tendon strain = $\epsilon'$	4	3	8	5	18	12

DD 907738/N/C

APPENDIX B

NUMBER OF PCRV HEAD TENDONS

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8.1 Yield Line Mode of Failure

The analysis is performed for the bottom head subjected to 1 RP or 1.5 RP cavity pressure. Calculational steps parallel those used in the FSAR Update, Section E. 11.2.2 (Ref. 1).

8.1.1 Basic Assumptions

See Ref. 1, Section. E. 11.2.2-2

Additional assumptions:

- 1) No vertical tendons.
- 2) Number of crosshead and circumferential tendons in the heads are to be reduced (Original design has 24 crosshead tendons and 34 circumferential tendons.)

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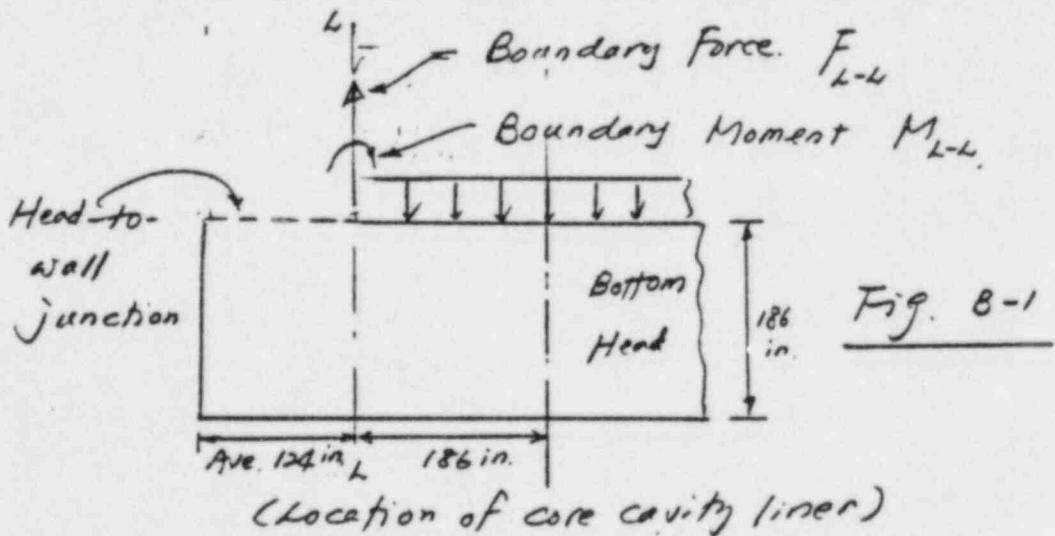
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B. 1.2 Calculational Procedure

- 1) Calculate the net head pressure load.  
 (Cavity pressure - cavity pressure equivalent)  
 of crosshead tendons

This is done for an assumed  $N_x$  (§B.1.3)

- 2) Calculate the net tensile force (or boundary force) transmitted through the intersection of the pcrv wall and the bottom head. (§B.1.4)



- 3) Assuming that a plastic hinge forms at the head-to-wall junction, calculate the boundary moment corresponding to  $F_{L-L}$ . (§B.1.5)

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- 1
- 2
- 3 4) Assuming a yield pattern for the head,  
5 calculate the unit yield line moment  
6 required to balance the net pressure.  
7  $F_{L-L}$  and  $M_{L-L}$ . (§B.1.6)
- 8
- 9
- 10 5) Calculate the number of circumferential  
11 (head) tendons required to provide a  
12 unit moment capacity along the yield  
13 equals to or larger than the required  
14 unit yield line moment. (§B.1.7. B.1.8)
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## CALCULATION SHEET

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B.1.3 Net Head Pressure Load

The pressure load on the heads is resisted in part by the crosshead tendons. The vertical component of the crosshead tendons is:

$$F_v = (2N_x) \cdot (A_t) \cdot (f_t) \sin \alpha$$

where:  $N_x$  = Number of crosshead tendons.

$A_t$  = X-sectional area of a tendon = .35 in.<sup>2</sup>

$f_t$  = tendon stress

$\alpha$  = average inclination of the tendons  
=  $36^\circ 45'$

Allowing  $f_t = 0.7 f_s' (1 - 0.12)(0.9) = 0.554 f_s' = 135$  ps.

$$F_v = 2N_x (0.35) (135,000) \sin 36^\circ 45'$$

$$= 1,329,000 N_x \text{ lbs.}$$

Cavity pressure equivalent of crosshead tendons is assumed constant over the cavity.

$$\frac{p}{\rho} = \frac{F_v}{3.14 \times 18^2} = 12.23 N_x \text{ psig.} \quad (\text{cavity rad.} = 18 \text{ in.})$$

## CALCULATION SHEET

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The net pressure load is:

$$P = \text{Actual cavity pressure} - 12.23 \text{ Nx}$$

Table B-1 Net Pressure and Boundary Force

- Nx	1.0 RP		1.5 RP	
	Net pressure (psi)	F <sub>L-L</sub> (k/in. of liner)	Net pressure (psi)	F <sub>L-L</sub> (k/in. of liner)
0	845	78.6	1268	117.9
6	772	71.8	1194	111.1
12	698	64.9	1120	104.2
18	625	58.1	1046	97.3
24	551	51.3	972	90.4

## CALCULATION SHEET

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B. 1.4 Boundary Force

Since there is no vertical tendon, the net pressure load must be resisted by the vertical rebars, liners and concrete.

The net tensile force per circumferential inch at the head-to-wall junction is:

$$F_{L-4} = \frac{\pi (186)^2 (\text{Net pressure in psi})}{2 \pi (186) (1000)} \quad k/\text{in. of liner,}$$

(core cavity radius)

See Table B-1.

## CALCULATION SHEET

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B. 1.5 Boundary Moment

Boundary moment at the head-to-wall junction is determined by establishing a strain diagram which is consistent with the plastic hinge condition and which will result in a net cross-section resistance force equals to  $F_{L-L}$ .

A plastic hinge is considered formed when,

$$\text{Max. concrete strain} = 0.003$$

or

$$\text{Max. steel strain (rebar)} = 0.07 = \epsilon'$$

Liner contribution is neglected in this analysis.

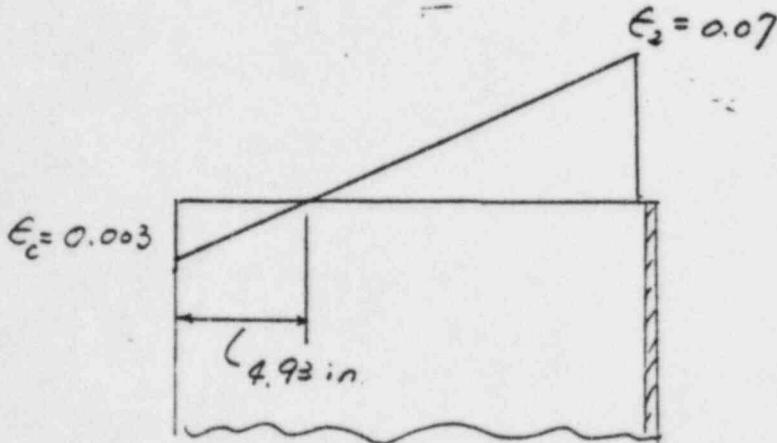
Fig. B-2 shows the location of rebars strain diagram, etc. for a case in which the two limits given above occur simultaneously.

## CALCULATION SHEET

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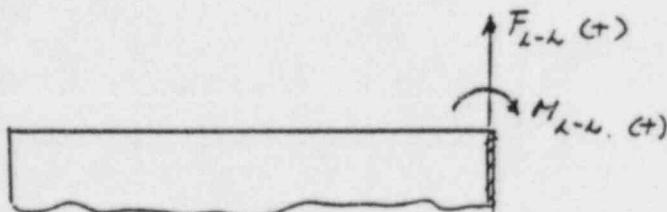
A balanced case:



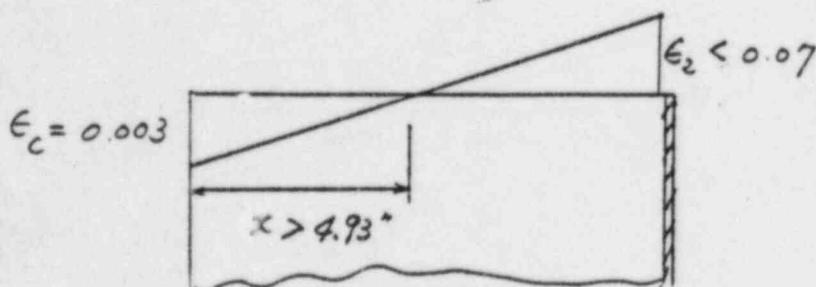
Referring to Fig. 8-2  
and Table B-2:

$$F_{L-L} = 76.6 \text{ k/in. of liner}$$

$$M_{L-L} = 1616.06 \text{ k-in./in. of liner}$$



For  $F_{L-L} < 76.6 \text{ k/in.}$ :



Case 1

T. T. Lee

11/19/74

907738 N/C

510

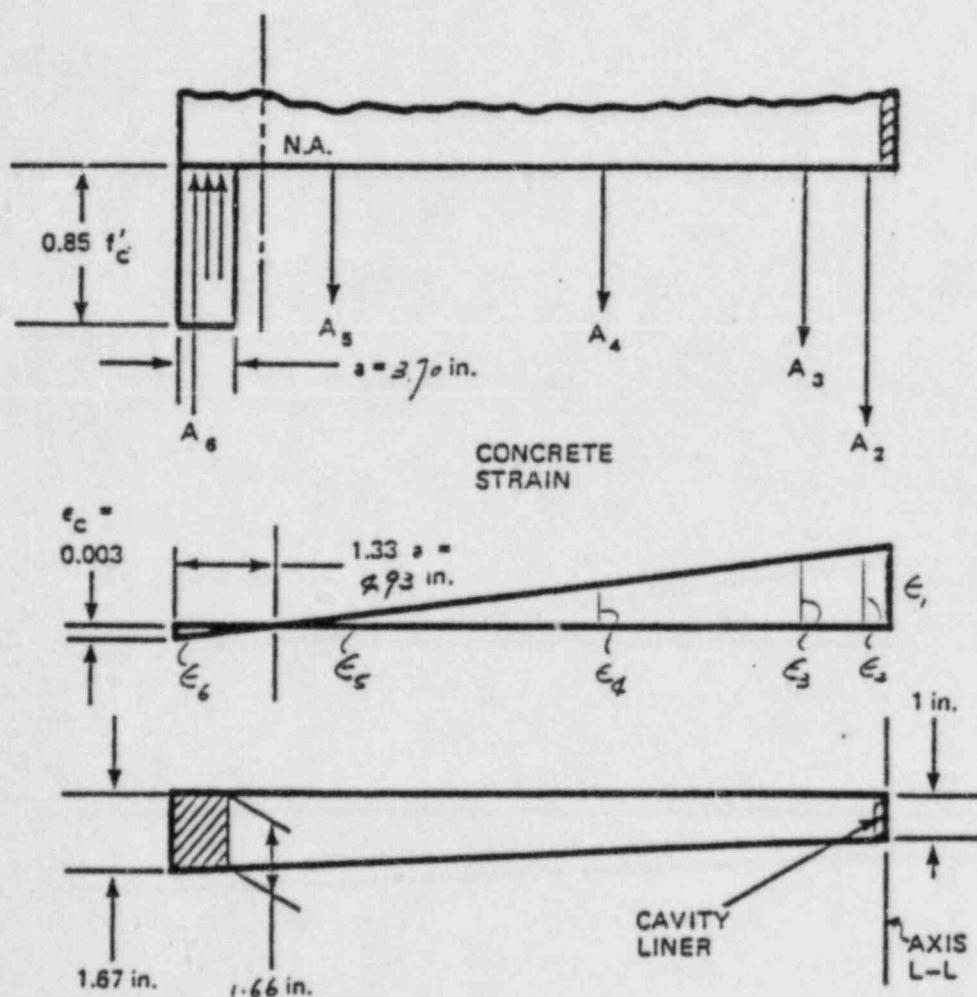
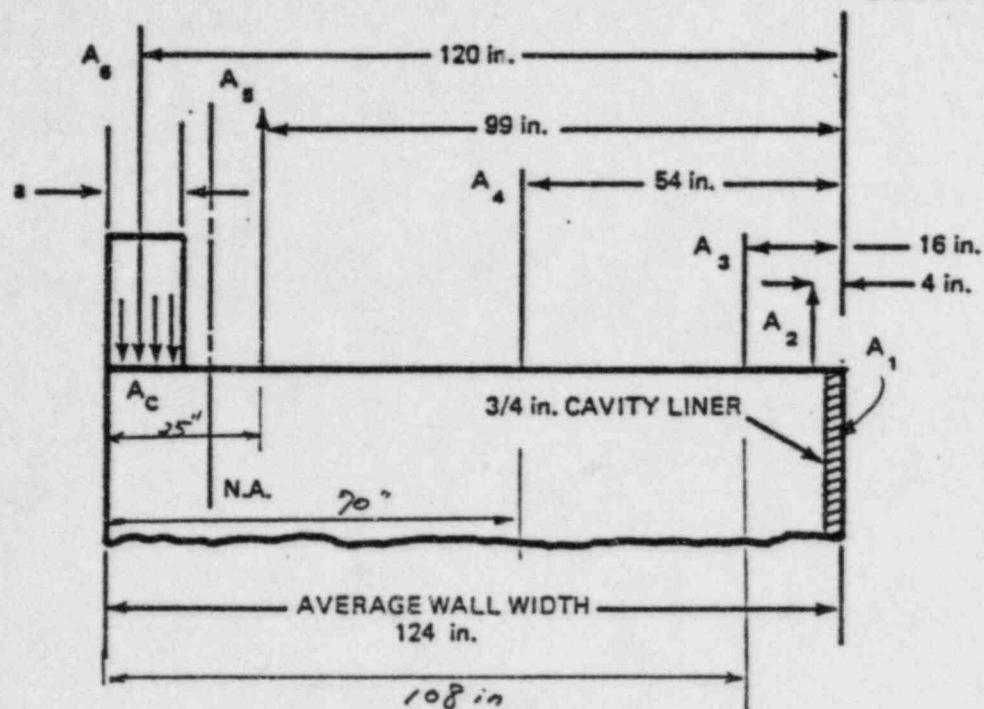


Figure B-2

## CALCULATION SHEET

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Table B-2  
Wall Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75		~	~	0	~
A <sub>2</sub>	0.27	0.07	90.	27.300	4.0	97.20
A <sub>3</sub>	0.29	0.063	82.640	25.126	16.0	402.02
A <sub>4</sub>	0.21	0.040	75.969	15.953	520	861.66
A <sub>5</sub>	0.77	0.012	63.331	48.765	99.0	4827.74
A <sub>6</sub>	0.28	-0.001	-16.412	-6.232	120.0	-748.32
A <sub>c</sub>		0.003		-3/.306	122.15	-3824.03

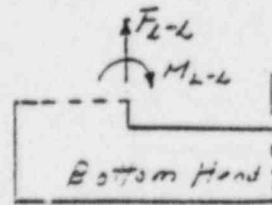
(a) Those for A<sub>2</sub> through A<sub>6</sub>

are from Ref. 1, Table E.1H1

(b) Based on x = 4.93 in.

$$\epsilon_c = 0.003$$

$$F_{L-L} = 76.602 \text{ (vs. )}, M_{L-L} = +161606$$



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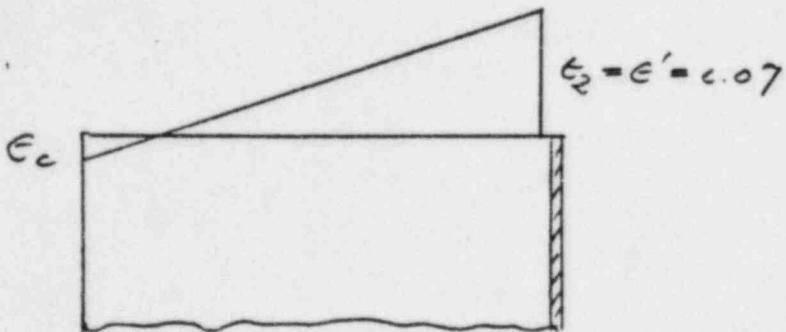
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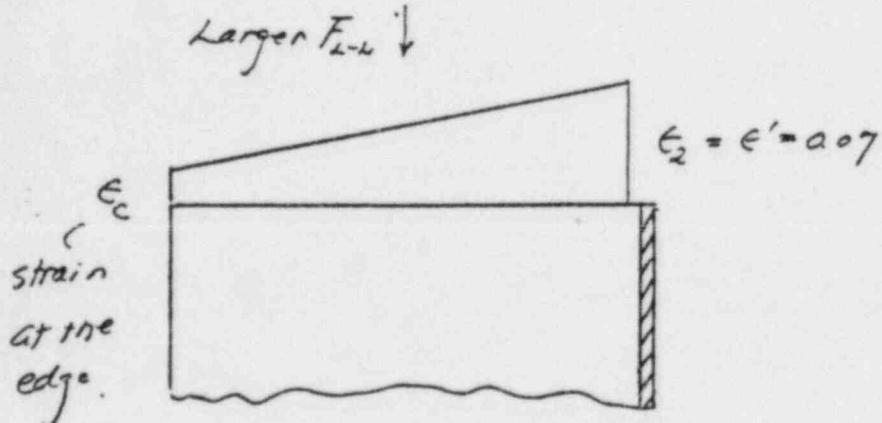
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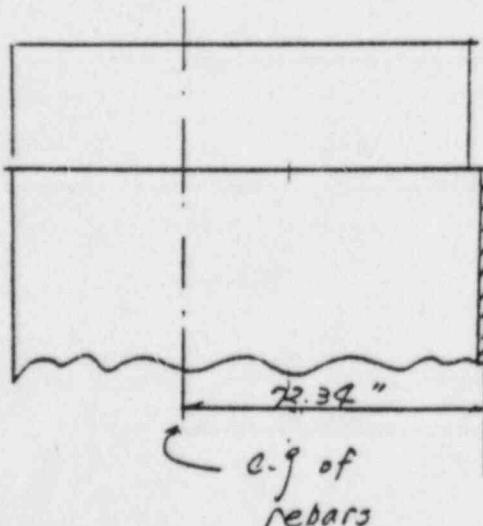
For  $F_{L-L} > 76.6 \text{ k/in}$ :



Case 2



Larger  $F_{L-L} \downarrow$



$$\epsilon = \epsilon' = 0.07 \text{ for all.}$$

$$F_{c,g} = f_s' \cdot Z A_s$$

$$= 90 \times 1.92$$

$$= 172.80 \text{ k/in}$$

$$M_{c,g} = 0$$

$$F_{L-L} = 172.80$$

$$M_{L-L} = +12500 \text{ k-in/in.}$$

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3 For Case 1,  $F_{L-L} < 766 \text{ k/in}$ :

4

5

6 From the force equilibrium:

7

8

9

$$\sum_{i=2}^5 F_i - F_c - F_c = F_{L-L} = \text{Boundary Force}$$

10

Note: Liner force not included.

11

Strains:

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14

$$\epsilon_1 = \frac{0.003(124-x)}{x} = \frac{0.372}{x} - 0.003$$

15

16

$$\epsilon_2 = \frac{0.003(120-x)}{x} = \frac{0.36}{x} - 0.003$$

17

18

$$\epsilon_3 = \frac{0.003(124-16-x)}{x} = \frac{0.003(108-x)}{x} = \frac{0.312}{x} - 0.003$$

19

20

$$\epsilon_4 = \frac{0.003(124-54-x)}{x} = \frac{0.003(70-x)}{x} = \frac{0.21}{x} - 0.003$$

21

22

$$\epsilon_5 = \frac{0.003(124-99-x)}{x} = \frac{0.003(25-x)}{x} = \frac{0.075}{x} - 0.003$$

23

24

25

$$\epsilon_6 = \frac{0.003(x-4)}{x} = 0.003 - \frac{0.012}{x} \quad (\text{comp.})$$

26

27

28

Stress:

Let  $f_i$  be the stress in element  $i$ .

29

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$$f_i = 90 - (0.07 - \epsilon_i) \times 461.5 \quad \text{if } \epsilon_i \geq 0.00202 \\ i=2 \sim 6 \quad (\text{Fig. A-2})$$

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$$f_6 = 29000 \epsilon_6 \quad \text{if } \epsilon_6 \leq 0.00202 \quad (\text{Fig. A-3})$$

$$f_c = 0.85 f'_c = 5.1 \text{ ksi}$$

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## Forces:

$$F_i = f_i \times A_i \quad i=2, \dots, 6$$

$$F_c = f_c \times A_c$$

$$\text{where } A_c = (0.75 z) (\text{Ave width of conc. compression zone}) \\ = (0.75 z) (\bar{w}_c)$$

$$\bar{w}_c = \frac{1}{2} (1.67 + 1.67 - 0.67 \cdot \frac{0.75x}{124})$$

$$= 1.67 - 0.00203x$$

$$\text{use } \omega_c = 1.66 \quad x \leq 7.29^\circ$$

$$a) = 1.65 \quad x = 12.31^\circ$$

$$w = 1.64 \quad x \leq 17.24$$

$$F_c = 6.35x \quad \text{if } x \leq 7.39''$$

$$= 6.3/x \quad \text{if } x = 12.3/''$$

$$= 6.27 x \quad \text{if } x \leq 17.24$$

Distance from L.L. for  $A_c$ :

$$12d - \frac{0.75x}{2} = 12d - 0.375x$$

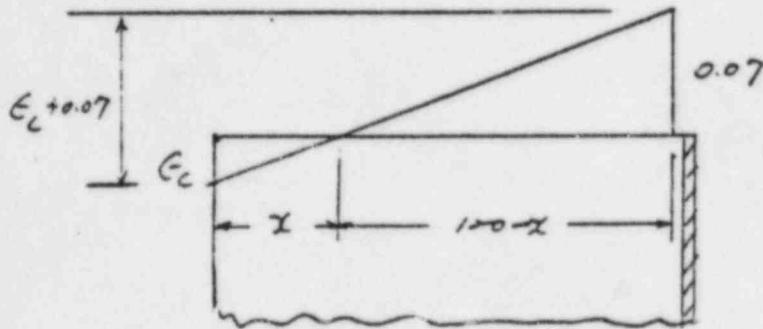
Calculation of  $M_{2-2}$  using the above formulas and other needed relations are done in tabular form.

## CALCULATION SHEET

## CALCULATIONS FOR

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Case 2: For  $F_{c-L} > 76.6 \text{ k/in}$



$$\epsilon_3 = 0.07 - (\epsilon_c + 0.07) \frac{12}{120} = 0.0630 - 0.1 \epsilon_c$$

$$\epsilon_4 = 0.07 - (\epsilon_c + 0.07) \frac{50}{120} = 0.0208 - 0.4167 \epsilon_c$$

$$\epsilon_5 = 0.07 - (\epsilon_c + 0.07) \frac{95}{120} = 0.0166 - 0.7917 \epsilon_c$$

$$\epsilon_6 = 0.07 - (\epsilon_c + 0.07) \frac{116}{120} = 0.0023 - 0.9667 \epsilon_c$$

$$x = \frac{120 \epsilon_c}{0.07 + \epsilon_c}$$

$$\text{use } F_c = 635x$$

$$+ arm = 124 - 0.375x \quad \} \text{ For } \epsilon_c \geq 0.001$$

(i.e., assume 0.85 f'\_c rect. block)

## CALCULATION SHEET

## CALCULATIONS FOR

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To avoid a tedious iteration process, calculation of  $M_{L-L}$  for various  $F_{L-L}$  given in Table B-1 is done as follows:

- 1) Compute a series of  $F_{L-L}$ ,  $M_{L-L}$  pairs for selected values of  $\lambda$  or  $\epsilon$ .  
Use formulas for Case 1 or Case 2, as needed  
(Tables B-2 through B-13)
- 2) Interpolate to obtain  $M_{L-L}$  for various  $F_{L-L}$  of interest.

(Table B-14)

## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-3 Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75	0.051	—	—	0	—
A <sub>2</sub>	0.27	0.029	80.389	21.705	20	86.820
A <sub>3</sub>	0.29	0.022	77.981	22.614	16.0	261.852
A <sub>4</sub>	0.21	0.027	70.356	14.775	520	797.829
A <sub>5</sub>	0.77	0.008	61.327	47.222	99.0	4674.972
A <sub>6</sub>	0.38	0.001	-36.565	-13.895	120.0	-1667.374
A <sub>c</sub>		0.003		-43.815	121.413	-5319.689

(a) Those for A<sub>2</sub> through A<sub>6</sub>

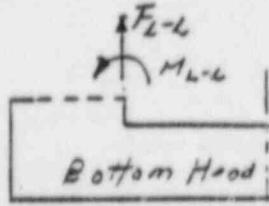
$$F_{L-L} = 48.61$$

$$M_{L-L} = -1065.63$$

are from Ref. 1, Table E.1H1

(b) Based on Z = 6.9 in.

$$\epsilon_c =$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-4 Boundary Force and  
Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (k/in. liner)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in/in.)
A <sub>1</sub>	0.75	0.055	—	—	0	—
A <sub>2</sub>	0.27	0.053	82.370	22.213	8.0	88.851
A <sub>3</sub>	0.29	0.048	79.674	23.105	16.0	369.620
A <sub>4</sub>	0.21	0.030	71.453	15.005	52.0	810.282
A <sub>5</sub>	0.77	0.009	61.719	47.523	99.0	2704.817
A <sub>6</sub>	0.38	0.001	-32.625	-12.398	120.0	-1487.700
A <sub>c</sub>		0.003		-40.640	121.6	-4981.824

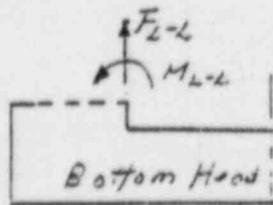
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 54.85 \quad M_{L-L} = -455.894$$

are from Ref. 1, Table E.11-1

(b) Based on X = 6.4 in.

$$E_c =$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-5 Boundary Force and  
Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in/in.)
A <sub>1</sub>	0.75	0.060	—	—	0	—
A <sub>2</sub>	0.27	0.058	84.375	22.781	4.0	91.125
A <sub>3</sub>	0.29	0.052	81.568	23.655	16.0	378.477
A <sub>4</sub>	0.21	0.032	72.681	15.263	52.0	824.206
A <sub>5</sub>	0.77	0.010	62.157	47.861	99.0	4730.221
A <sub>6</sub>	0.38	0.001	-28.216	-10.722	120.0	-1286.659
A <sub>c</sub>		0.003		-37.592	121.78	-4577.957

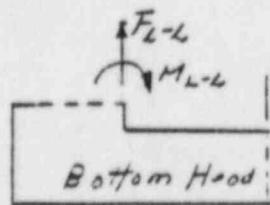
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 61.25 \quad M_{L-L} = 167.539$$

are from Ref. 1, Table E.11-1.

(b) Based on x = 5.92 in.

$$\epsilon_c =$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-6 Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in/in.)
A <sub>1</sub>	0.75	0.065	—	—	0	—
A <sub>2</sub>	0.27	0.062	86.52	23.36	70	93.4
A <sub>3</sub>	0.29	0.056	83.50	24.21	16.0	387.4
A <sub>4</sub>	0.21	0.035	73.93	15.53	560	838.4
A <sub>5</sub>	0.77	0.011	62.60	18.20	99.0	4772.3
A <sub>6</sub>	0.38	-0.001	-23.73	-9.02	120.0	-1082.0
A <sub>c</sub>		0.003		-34.92	121.94	-4258.7

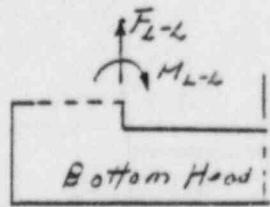
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 67.36 \quad M_{L-L} = +750.8$$

are from Ref. 1, Table E.1H-1

(b) Based on x = 5.5 in.

$$\epsilon_c = 0.003$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-7. Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in. <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	8.0	97.2
A <sub>3</sub>	0.29	0.0627	86.64	25.12	16.0	402.0
A <sub>4</sub>	0.21	0.0396	75.97	15.95	52.0	861.5
A <sub>5</sub>	0.77	0.0123	63.37	48.80	99.0	4830.8
A <sub>6</sub>	0.38	-0.0005	-14.60	-5.55	120.0	-665.7
A <sub>c</sub>		0.0029		-30.29	122.21	-3701.7

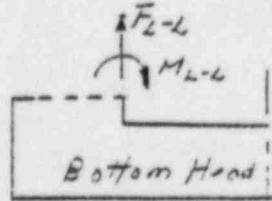
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 78.3 \quad M_{L-L} = 1824.1$$

are from Ref. 1, Table E.1K1.

(b) Based on x = 4.77 in.

$$\epsilon_c = 0.0029$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-8 Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in/in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	4.0	97.2
A <sub>3</sub>	0.29	0.0628	86.66	25.13	16.0	402.1
A <sub>4</sub>	0.21	0.0398	76.08	15.98	520	862.8
A <sub>5</sub>	0.77	0.0128	63.59	48.97	99.0	4847.7
A <sub>6</sub>	0.28	0.0001	2.22	0.84	120.0	101.3
A <sub>c</sub>		0.0023		-24.24	122.57	-2971.0

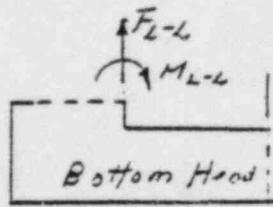
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 90.98 \quad M_{L-L} = 3340.0$$

are from Ref. 1, Table E.1H!

(b) Based on x = 3.82 in.

$$E_c = 0.0023$$



## CALCULATION SHEET

## CALCULATIONS FOR

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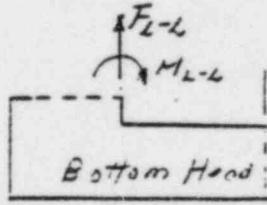
Table B-9 Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	7.0	97.2
A <sub>3</sub>	0.29	0.0628	86.68	25.14	16.0	402.2
A <sub>4</sub>	0.21	0.0400	76.14	15.99	520	863.4
A <sub>5</sub>	0.17	0.0130	63.70	49.05	99.0	4856.0
A <sub>6</sub>	0.28	0.0004	11.60	4.41	120.0	529.0
A <sub>c</sub>		0.002		-21.17	122.75	-2591.6

(a) Those for A<sub>2</sub> through A<sub>6</sub>       $F_{L-L} = 97.72$        $M_{L-L} = 4149.2$   
 are from Ref. 1, Table E.11-1.

(b) Based on  $x = 333$  in.

$$\epsilon_c = 0.002$$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-10 Boundary Force and Moment about Liner Axis L-L

Element	Area <sup>(a)</sup> in <sup>2</sup> /in)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	70	972
A <sub>3</sub>	0.29	0.0628	86.69	25.14	16.0	402.2
A <sub>4</sub>	0.21	0.0401	76.20	16.00	570	8641
A <sub>5</sub>	0.77	0.0133	63.81	29.14	99.0	48644
A <sub>6</sub>	0.28	0.0007	19.04	7.24	120.0	1027.6
A <sub>c</sub>		0.0017		-18.07	122.9	-2221.4

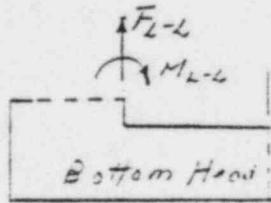
(a) Those for A<sub>2</sub> through A<sub>6</sub>

$$F_{L-L} = 103.75 \quad M_{L-L} = 50551$$

are from Ref. 1, Table E.11-1

(b) Based on x = 2.84 in.

$$\epsilon_c = 0.0017$$



## CALCULATION SHEET

## CALCULATIONS FOR

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1  
2 Table B-11 Boundary Force and  
3 Moment about Liner Axis L-L  
4  
5

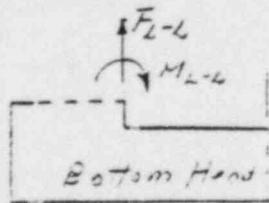
Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in./in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	8.0	97.2
A <sub>3</sub>	0.29	0.0629	86.71	25.14	16.0	402.3
A <sub>4</sub>	0.21	0.0403	76.27	16.02	57.0	869.9
A <sub>5</sub>	0.77	0.0136	63.96	29.25	99.0	4875.5
A <sub>6</sub>	0.38	0.0010	30.26	11.50	120.0	1379.6
A <sub>c</sub>		0.0013		-13.89	123.18	-1710.9

30 (a) Those for A<sub>2</sub> through A<sub>6</sub>       $F_{L-L} = 112.32$        $M_{L-L} = +59 = ?$  6  
31

32 are from Ref. 1, Table E.11-1

33 (b) Based on x = 2.19 in.

34  $\epsilon_c = 0.0013$



## CALCULATION SHEET

## CALCULATIONS FOR

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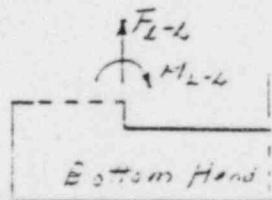
1  
2      Table B-12      Boundary Force and  
3      Moment about Liner Axis L-L  
4  
5

Element	Area <sup>(a)</sup> (in <sup>2</sup> /in)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt. L-L (k-in/in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90.0	24.3	7.0	97.2
A <sub>3</sub>	0.29	0.0629	86.723	25.15	16.0	402.2
A <sub>4</sub>	0.21	0.0404	76.34	16.03	52.0	865.7
A <sub>5</sub>	0.77	0.0138	64.07	49.33	99.0	2883.9
A <sub>6</sub>	0.38	0.0013	37.7	14.33	120.0	1719.1
A <sub>c</sub>		0.001		-10.73	123.37	-1323.7

30      (a) Those for A<sub>2</sub> through A<sub>6</sub>      F<sub>L-L</sub> = 117.65      M<sub>L-L</sub> = 664.5

31      are from Ref. 1, Table E.11-1.

32      (b) Based on x = 1.69 in.       $\epsilon_c = 0.001$



## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-13 Boundary Force and Moment about Liner Axis L-L

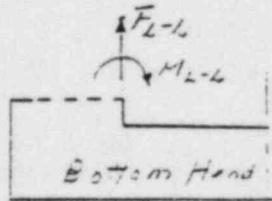
Element	Area <sup>(a)</sup> (in <sup>2</sup> /in.)	Strain <sup>(b)</sup> (in./in.)	Stress (ksi)	Force (k/in. liner)	Dist. from L-L <sup>(a)</sup> (in.)	Moment abt L-L (k-in/in.)
A <sub>1</sub>	0.75		—	—	0	—
A <sub>2</sub>	0.27	0.07	90	24.3	20	91.2
A <sub>3</sub>	0.29	0.0630	86.77	25.16	16.0	402.6
A <sub>4</sub>	0.21	0.0408	76.52	16.07	570	867.8
A <sub>5</sub>	0.77	0.0126	64.43	49.61	99.0	4911.7
A <sub>6</sub>	0.27	0.0023	58.76	22.33	120.0	2679.3
A <sub>c</sub>						

(a) Those for A<sub>2</sub> through A<sub>6</sub>  $F_{L-L} = 137.47$   $M_{L-L} = 8950.6$

are from Ref. 1, Table E.1b-1

(b) Based on  $X =$  in

$$\epsilon_c = 0$$



Bottom Flange

## CALCULATION SHEET

## CALCULATIONS FOR

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1 Table B-14  $M_{L-L}$  Values by Interpolation  
2

Calculated (See Tables )		Interpolated	
$F_{L-L}$	$M_{L-L}$	$F_{L-L}$	$M_{L-L}$
48.61	-1065.6	51.3	-802.8
54.85	-255.9	58.1	-139.4
61.25	167.4	64.9	515.9
67.35	750.8	71.8	1166.6
73.60	1616.1	78.6	1860.0
79.30	1824.1	85.4	3270.7
85.98	3340.0	92.3	4098.8
97.72	4149.2	104.2	5099.0
103.75	5058.1	111.1	5787.0
112.32	5908.6	117.9	6673.8
117.65	6644.6		
137.47	8958.6		

## CALCULATION SHEET

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B.1.6 Unit Yield Line Moment

Consider two failure modes as shown in Fig. E-3

For each mode, calculate the unit yield line moment required to maintain equilibrium.

Forces acting on each segment of slab bounded by the yield lines and the boundaries are:

- 1) Net cavity pressure
- 2) Boundary forces and moments
- 3) Yield line moment

Equilibrium of these forces are established by computing their moments about line c-c' (or E-E') shown in Fig. B-4.

The following calculation is performed for the case of 1.0 RP with  $N_x = 0$ . Unit yield moments for other cases are done by proportion from this basic case (Tables B-15 ~ B-18).

T. T. Lee "1/19/84

907738 N/C 831

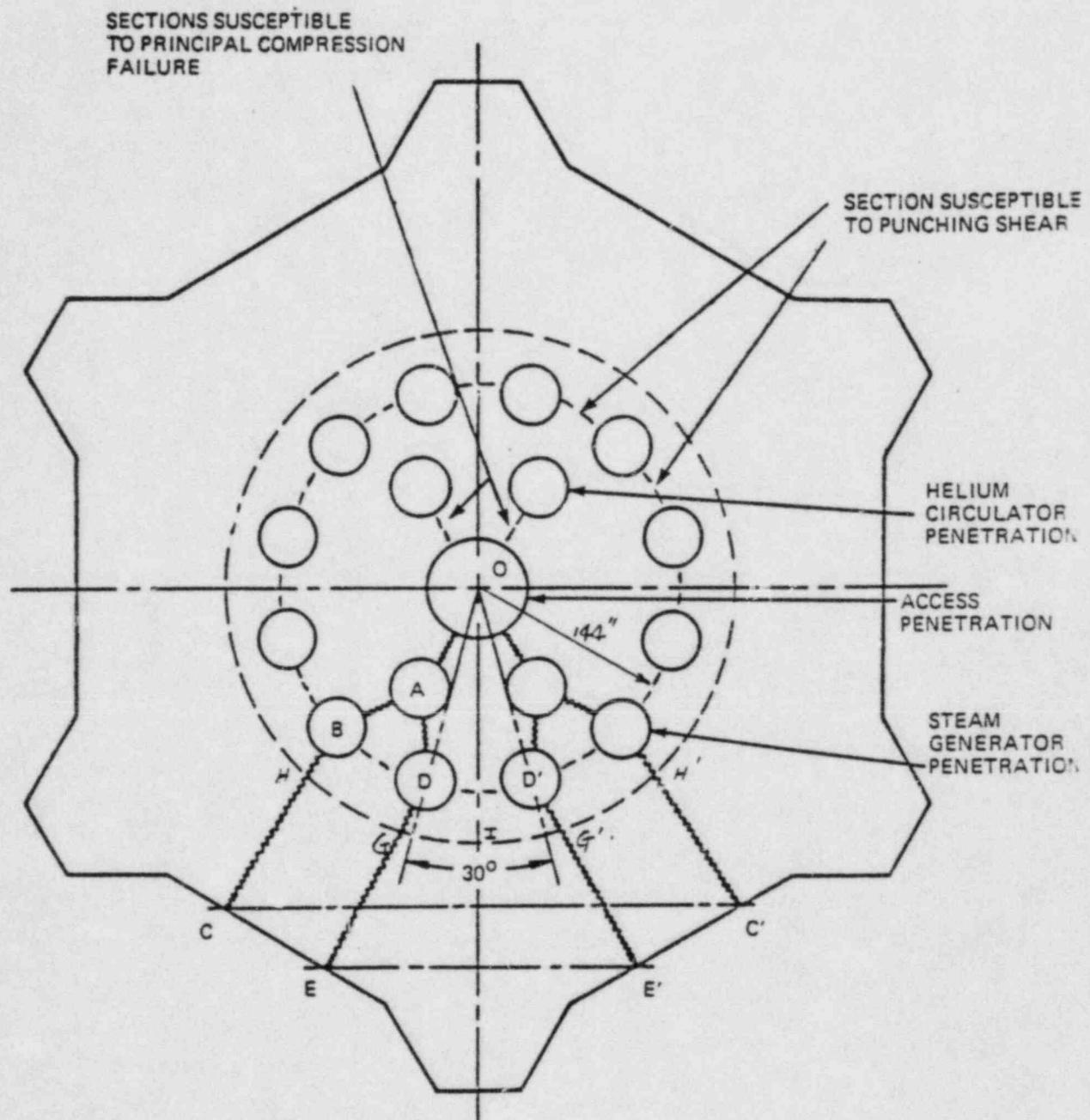


Figure B-3 Arrangement of Lower Head Penetrations and Potential Failure Modes

# CALCULATION SHEET

GA 268 (REV. 4/82)

## CALCULATIONS FOR

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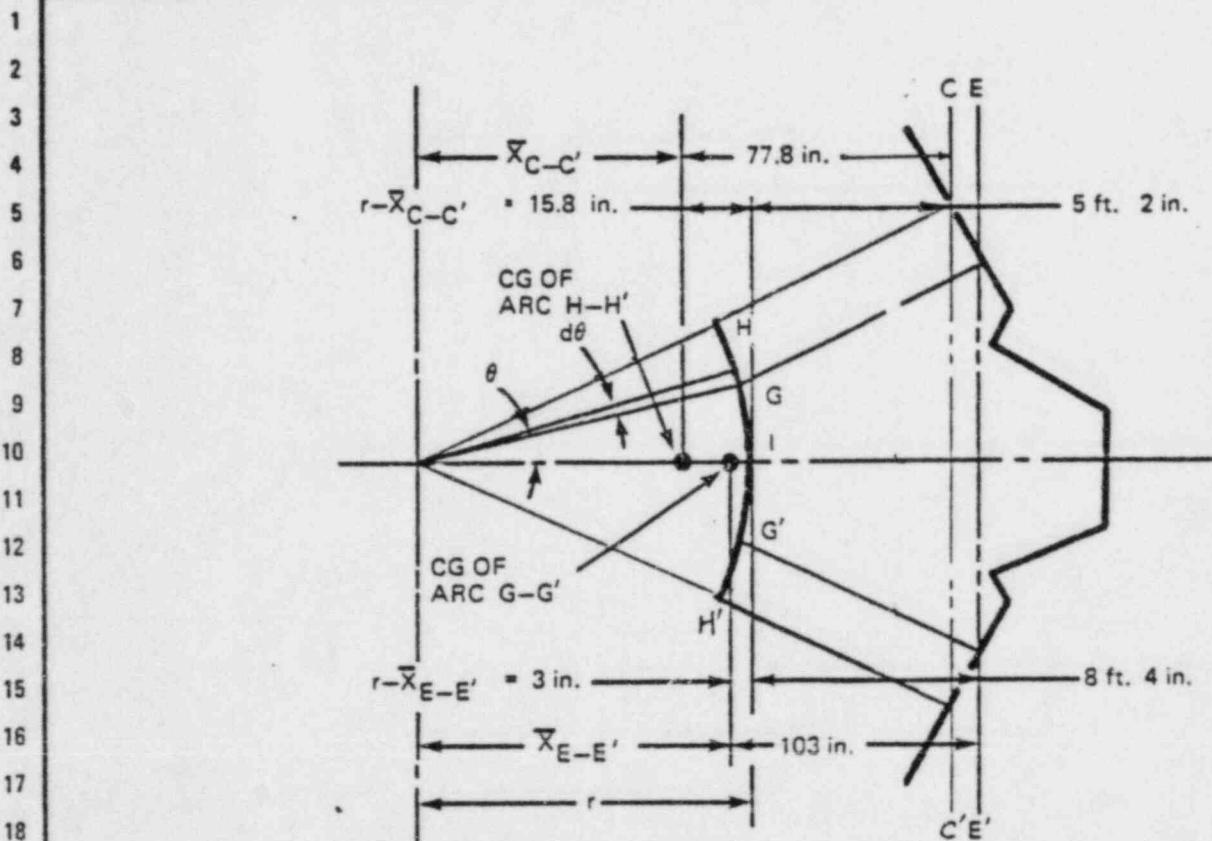


Fig. B-4

## CALCULATION SHEET

## CALCULATIONS FOR

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3 Yield pattern O-A-B-C (See Fig. B-3):

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Referring to Fig. B-5, the moment about Line A-A' due to the uniform distributed moment  $m$  is:

$$M_{A-A'} = 2 \int_0^{\theta_1} mr \cos \theta d\theta = 2mr [\sin \theta]_0^{\theta_1}$$

$$= 2mr \sin \theta_1$$

For this case,  $\theta_1 = 41^\circ 30'$  (Fig. B-6)

Hence, due to  $M_{L-L}$  (See Table B-1a):

$$M_{C-C'} = M_{H-H'} = 2(1860 \times 1000)(186) \sin 41^\circ 30'$$

$$= 0.458 \times 10^9 \text{ lbs-in}$$

Due to  $F_{L-L}$  over H-H'.

$$M_{C-C'} = 78.6 \times 10^3 \times \frac{2 \times 41.5 \times \pi}{180} \times 186 \times 77.8$$

$$= 1.648 \times 10^9 \text{ lbs-in}$$

where 77.8 in. is the distance between the c.g. of  $F_{L-L}$  over H-H' and Line C-C' (See Fig. B-4).

# CALCULATION SHEET

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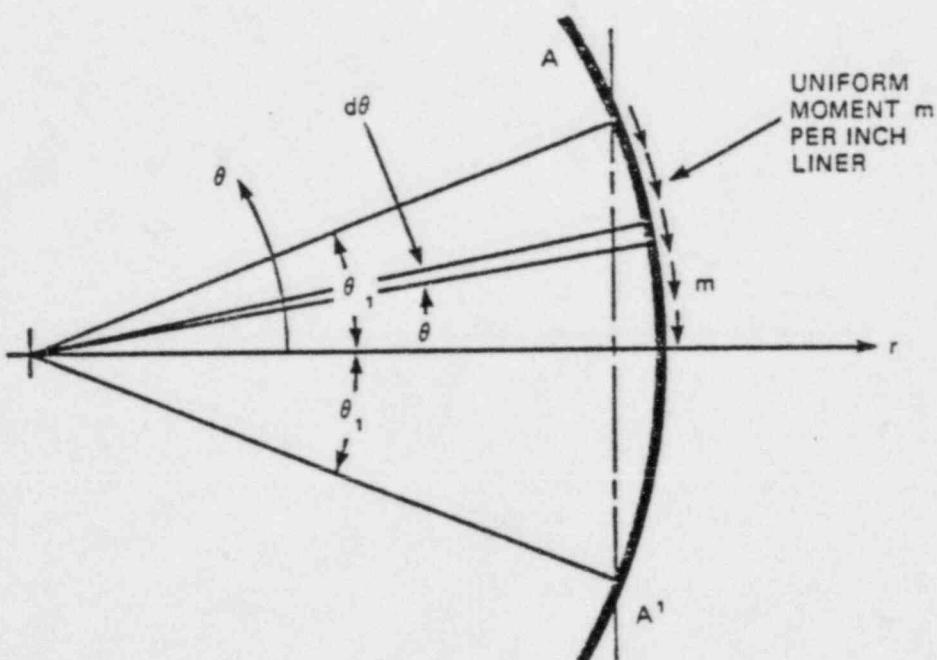
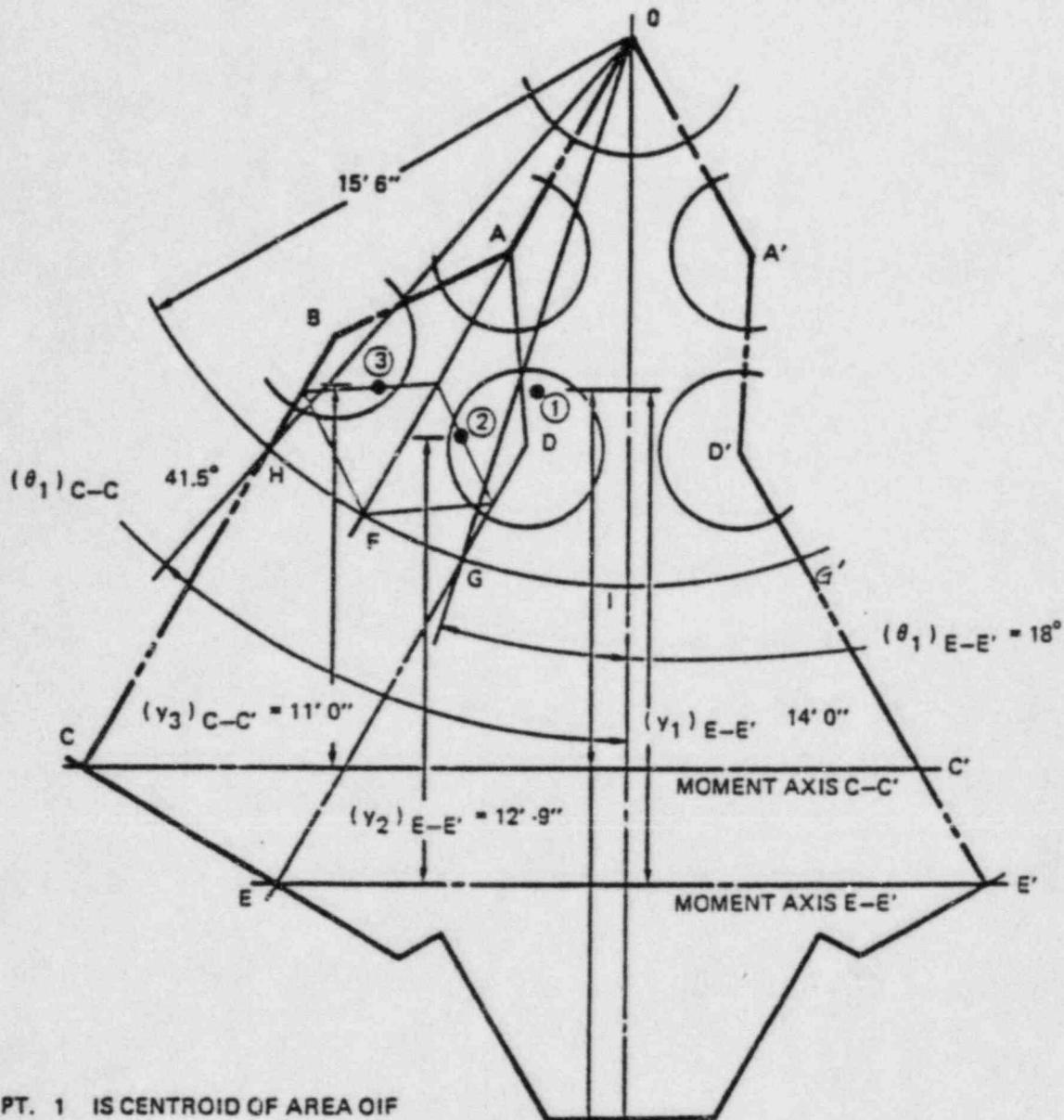


Fig. B-5

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NOTE: PT. 1 IS CENTROID OF AREA OIF

PT. 2 IS CENTROID OF AREA ADGF

PT. 3 IS CENTROID OF AREA ABHF

$(y_1)_{C-C'} = 10' 10''$

$(y_1)_{E-E'} = 14' 0''$

$(y_2)_{E-E'} = 12' 9''$

$(y_3)_{C-C'} = 11' 0''$

$(\theta_1)$

$C-C'$

$41.5^\circ$

$(\theta_1)_{E-E'} = 18^\circ$

Fig. B-6

Potential Failure Mode

## CALCULATION SHEET

## CALCULATIONS FOR

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3 Referring to Fig. 10, and Ref. & P.E. 11-7,

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$$\text{Area OAFI} = \frac{1}{2} \pi (186)^2 = 9050 \text{ in}^2$$

6

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8

$$\text{Centroidal dist. from C-C'} = 10' 10'' = 130 \text{ in}$$

9

10

$$\text{Area ABHF} = \frac{1}{2} (102 + 42) \times 38 = 2736 \text{ in}^2$$

11

12

$$\text{Centroidal dist. from C-C'} = 11' = 132 \text{ in}$$

13

14

Hence moment about C-C'. due to the

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cavity pressure on (OABHI)  $\times 2$  is:

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18

$$M_{C-C'} = 2 p [(9050 \times 130) + (2736 \times 132)]$$

19

20

$$= 3.076 p \times 10^6 \text{ lbs-in}$$

21

22

$$= 3.076 \times 845 \times 10^6 = 2.599 \times 10^9 \text{ lbs-in}$$

23

24

Total moment about C-C' to be resisted  
by the yield lines O-A-B-C is:

25

26

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28

$$M_{C-C'} = (2.599 - 1.648 + 0.458) \times 10^9$$

29

30

$$= 1.410 \times 10^9 \text{ lbs-in}$$

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## CALCULATION SHEET

## CALCULATIONS FOR

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Let  $m_{B-C}$  be the unit yield line moment for all segments along O-A-B-C.

The components of total yield line moment for individual segments in the C-C' direction is (see Fig. B-7):

$$2 \left( 17 m_{B-C} \cdot \frac{9}{17} + 14 m_{B-C} \cdot \frac{13}{14} + 19_2 m_{B-C} \cdot \frac{22}{19_2} \right) \\ = 188 m_{B-C}$$

This balances the  $M_{C-C'}$  total, hence

$$m_{B-C} = \frac{M_{C-C'} \text{ total}}{188} = \frac{1.410 \times 10^9}{188} = 7.50 \times 10^6 \text{ lb-in.}$$

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907738 N/C B38

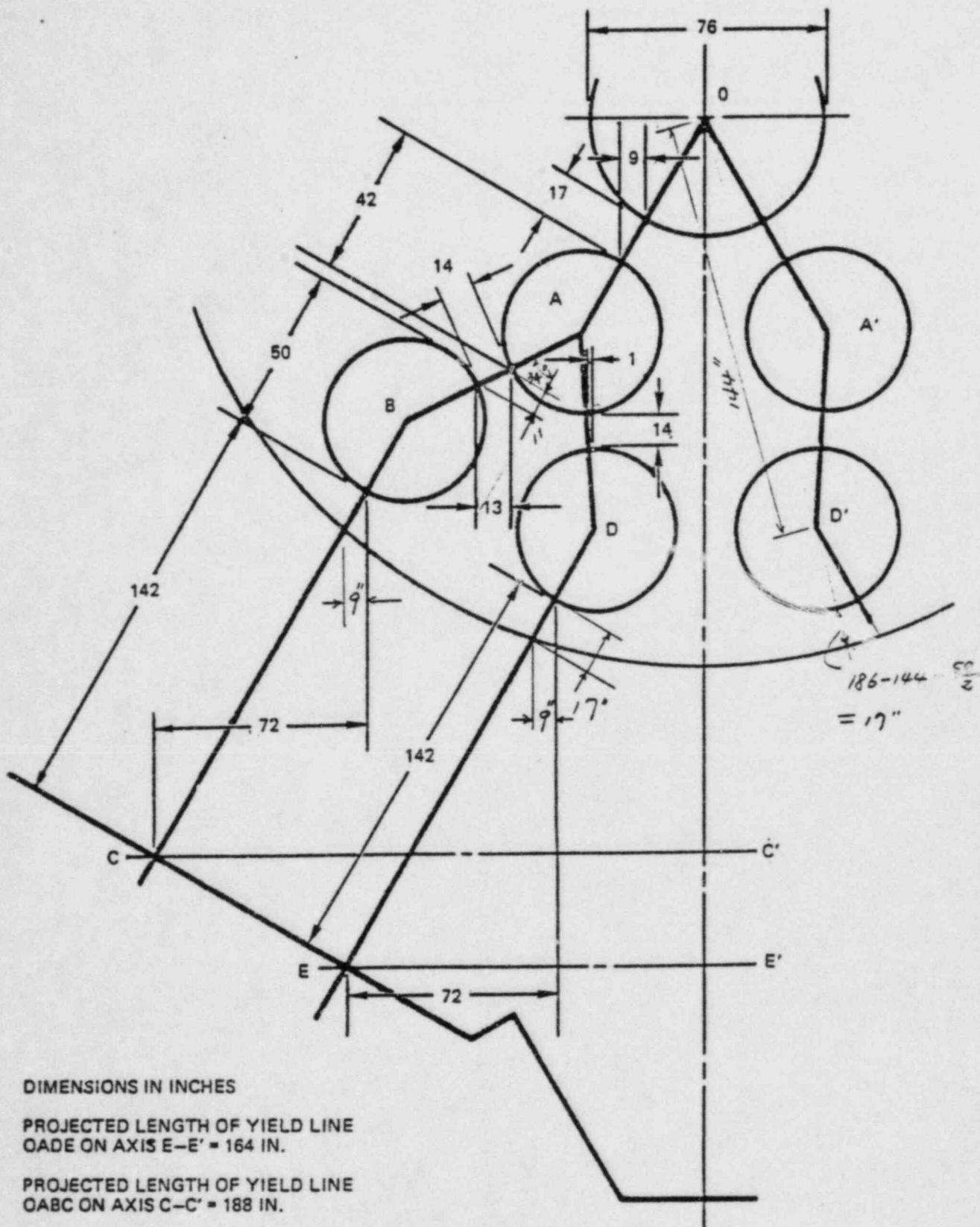


Figure 8-7 Projected Length of Yield Line

## CALCULATION SHEET

## CALCULATIONS FOR

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Yield pattern O-A-D-E:

Repeating the similar calculation done  
for yield pattern O-A-B-C.

Due to  $M_{L-L}$  over G-G'.

$$M_{E-E'} = 2(1860 \times 1000)(186) \sin 18^\circ = 0.214 \times 10^9$$

1bs-in

Due to  $F_{L-L}$  over G-G'.

$$M_{E-E'} = (78.6 \times 1000) \times \frac{2 \times 18 \times \pi}{180} \times 186 \times 103$$

$$= 0.946 \times 10^9 \text{ 1bs-in.}$$

Due to cavity pressure over (OADGI)  $\times 2$ :

$$M_{E-E'} = 2P [(9050 \times 168) - (2736 \times 153)]$$

$$= 2.204 P \times 10^6 \text{ 1bs-in}$$

$$= 2.204 \times 845 \times 10^6 = 1.862 \times 10^9 \text{ 1bs-in}$$

Total to be resisted by the yield lines

O-A-D-E is:

$$M_{E-E'} = (1.862 - 0.946 + 0.214) \times 10^9$$

$$= 1.130 \times 10^9 \text{ 1bs-in.}$$

## CALCULATION SHEET

## CALCULATIONS FOR

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Let  $m_{D-E}$  be the unit yield line moment for all segments along O-A-D-E.

The components of total yield line moments for individual segments along the E-E' axis is (Fig. B-7) :

$$\begin{aligned} & 2(9m_{D-E} + m_{D-E} + 72m_{D-E}) \\ & = 164m_{D-E} \end{aligned}$$

This balances the  $M_{E-E'}$  total, hence

$$m_{D-E} = \frac{M_{E-E', \text{total}}}{164} = \frac{1.130 \times 10^9}{164} = 6.89 \times 10^6 \text{ lbs-in/in}$$

The  $m_{B-C}$  of  $7.50 \times 10^6$  lbs-in/in and

the  $m_{D-E}$  of  $6.89 \times 10^6$  " computed for the case of 1 RP and  $N_x=0$  are used as the basis for proportioning in the following tables.

## CALCULATION SHEET

## CALCULATIONS FOR

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1

2 Table B-15

3

4 Unit Yield Line Moments

5

6 for Yield Pattern O-A-B-C, 1 RP

$N_x$	0	6	12	18	24
$M_{L-L}$	18600	11666	5159	-1394	-8022
$F_{L-L}$	78.6	71.8	64.9	58.1	51.3
Resul. Cavity Pressure	845	772	698	625	551
$M_{C-C}$	* $M_{L-L}$	0.458	0.287	0.127	-0.034
Due	* $F_{L-L}$	-1.648	-1.505	-1.361	-1.218
to	* ( $10^6$ lb-in.)	Press.	2.599	2.374	2.147
Total		1.410	1.156	0.913	0.670
$M_{C-C}$					0.421
$m_{B-C}$	* ( $10^6$ lb-in./in.)	7.50	6.15	4.86	3.56
Note:	↑ Base For Tables B-15 and B-17		* Numbers in these rows are computed by proportion from the base case in Table B-15		

## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-16

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Unit Yield Line Moments

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for Yield Pattern O-A-D-E, 1 RP

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$N_x$	0	6	12	18	24
$M_{L-L}$	1860.0	1166.6	515.9	-139.4	-802.?
$F_{L-L}$	126	718	64.9	58.1	51.3
Resul. Cavity Pressure	845	772	698	625	551
$M_{E-E}^{**}$	-0.214	0.134	0.059	-0.016	-0.092
$M_{E-E}^{**}$ Due to	-0.946	-0.864	-0.781	-0.699	-0.617
( $10^9$ lb-in) Press.	1.862	1.701	1.538	1.377	1.212
Total $M_{E-E}^{**}$	1.130	0.971	0.816	0.662	0.505
$m_{O-E}^{**}$ ( $10^6$ lb-in/in.)	6.89	5.92	4.98	4.07	3.08

Base Case

Note:

(For Tables B-16, B-18)

\*\* Numbers in these rows are computed by proportion from the base case in Table B-16.

## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-17

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Unit Yield Line Moments

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for Yield Pattern o-A-B-C, 1.5 RP

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$N_x$	0	6	12	18	24
$M_{L-L}$	6673.8	5787.0	5099.0	4098.8	3270.7
$F_{L-L}$	117.9	111.1	104.2	97.3	90.5
Resul. Cavity Pressure	1268	1194	1120	1046	972
$M_{C-C}^*$	1.643	1.425	1.256	1.009	0.805
Due to $F_{L-L}^*$	-2.272	-2.329	-2.185	-2.040	-1.895
( $10^9$ lb-in) Press.	3.900	3.672	3.425	3.217	2.990
Total $M_{C-C}^*$	3.071	2.768	2.516	2.186	1.900
$M_{B-C}^*$ ( $10^6$ lb-in/in)	16.34	14.72	13.38	11.63	10.11

$$M_{B-C} = 11.27$$

if  $N_x = 19.42$  by interpolation

## CALCULATION SHEET

## CALCULATIONS FOR

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Table B-18

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3 Unit Yield Line Moments

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5 for Yield Pattern O-A-D-E, 15 RP

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$N_x$	0	6	12	18	24
$M_{L-L}$	6673.8	5787.0	5099.0	4098.8	3270.7
$F_{L-L}$	117.9	111.1	109.2	97.3	90.4
Resul. Cavity Pressure	1268	1197	1120	1046	972
$M_{E-E}^{**}$	0.768	0.666	0.587	0.472	0.376
$M_{E-E}^{**}$ Due to Press.	-1419	-1337	-1254	-1171	-1088
Total $M_{E-E}^{**}$	2.143	1.960	1.801	1.606	1.430
$M_{D-E}^{**}$ ( $10^6$ lb-in/in.)	13.07	11.95	10.98	9.79	8.72

## CALCULATION SHEET

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B.1.7 Moment Capacity Along Yield Lines

Moment capacity in the hoop direction provided by the circumferential tendons, circumferential rebars and concrete is established for a cross section with unit width along a radial yield line.

Criteria:

- 1) Maximum tendon stress = 0.9  $f_{sy}$  (Ref. 1)
- 2) Maximum concrete strain  $\leq 0.003$

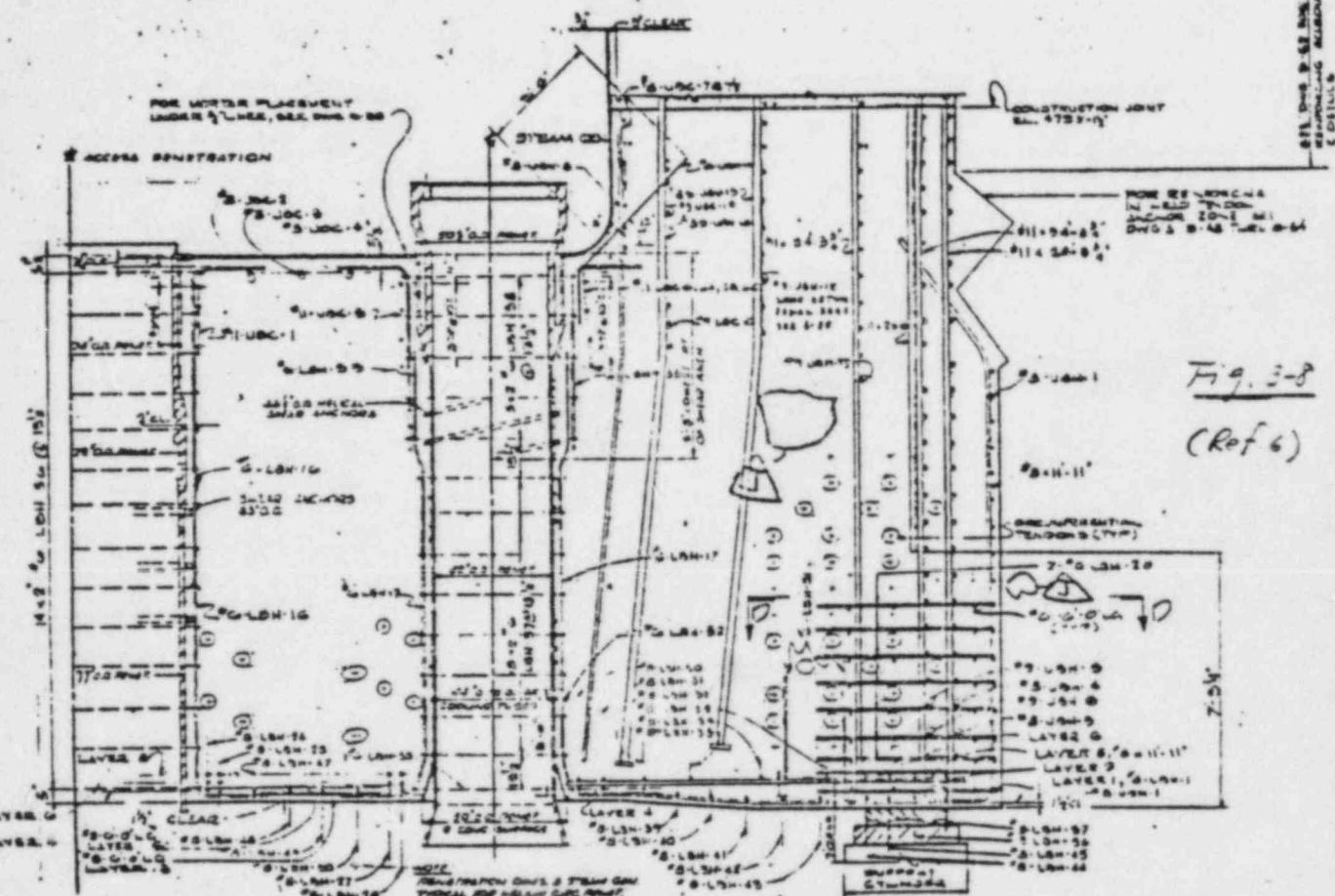
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### Circumferential Rebars:



For 17" long concrete ligament between O. and A (Fig. 2-1):

$$6 - \#8 \text{ bars} \rightarrow \frac{6 \times 0.79}{17} = 0.28 \text{ in}^2/\text{in}$$

For 12" long concrete ligament between A and D:

$$6 - \#8 \text{ bars} \rightarrow 0.28 \text{ in}^2/\text{in}$$

For 142" length between D & E:

$$22 - \#8 \text{ bars} \rightarrow \frac{22 \times 0.79}{142} = 0.12 \text{ in}^2/\text{in}$$

## CALCULATION SHEET

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Rebars: LBH - 30 (2), -31(2), -32(2)

3

(Fig. 8-8) LBH - 33 (1), -34(1), -35(1)

4

LBH - 40 (1), -41(1), -42(1)

5

LBH - 43 (1), -44(2), -45(2)

6

LBH - 56 (2), -57(2)

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Each contains three sections, forming a complete circle.

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use  $A_r = 0.12 \text{ in}^2/\text{in}$  in the following calculation.

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## CALCULATIONS FOR

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Effect of Internal Pressure within Penetration:

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For concrete compression zone:

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where  $P_c$  = cavity pressure. $x$  = height of compression zone

0.722 is from Ref. - P.E. 11-10.

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For tension zone:

Stress in penetration is:

$$\frac{P_c \cdot r}{h} = \frac{845 \times 21}{1} = 17,700 \text{ psi.}$$

$< f_{sy}$ .

 $r$  = radius of penetration = 21 in. $h$  = thickness of penetration = 1 in. } (Ref. 1.)Hence this effect is applied to compression  
zone only in the calculation of moment capacity.

## CALCULATION SHEET

## CALCULATIONS FOR

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Moment Capacity Per Unit Width Along Yield Line:

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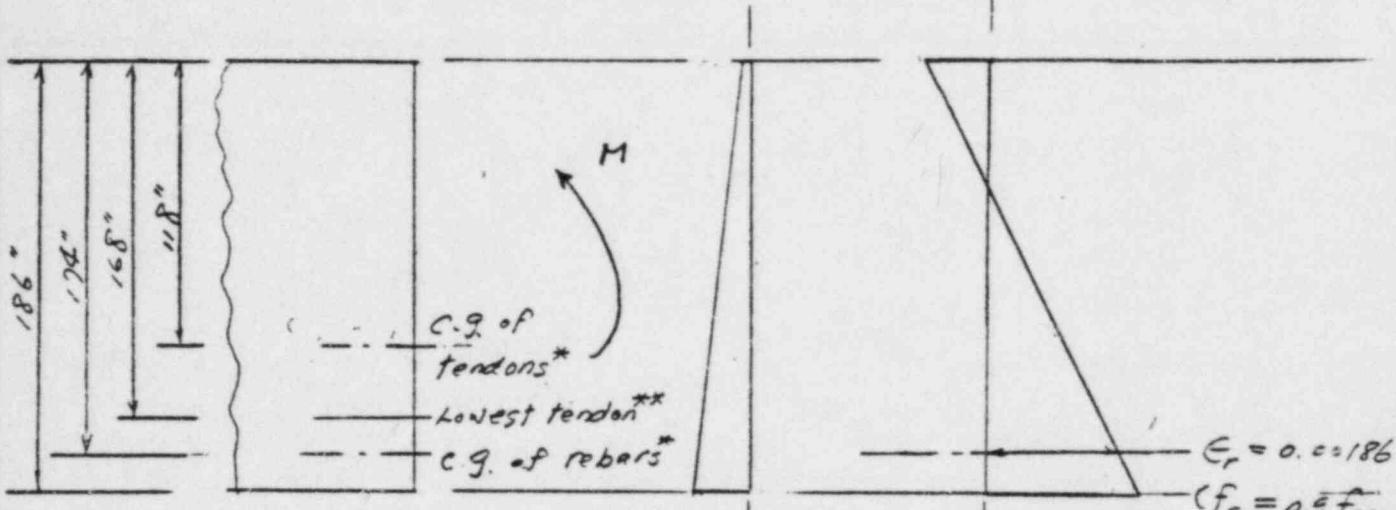
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\* Ref. 1, Fig. E-11-7

(1) strain due

(2) strain at

$$\epsilon_r = 0.00186$$

$$(f_{r,c} = 0.9 f_{s,y})$$

$$= 55 \text{ ksi}$$

\*\* Ref. 5.

to prestress

limit

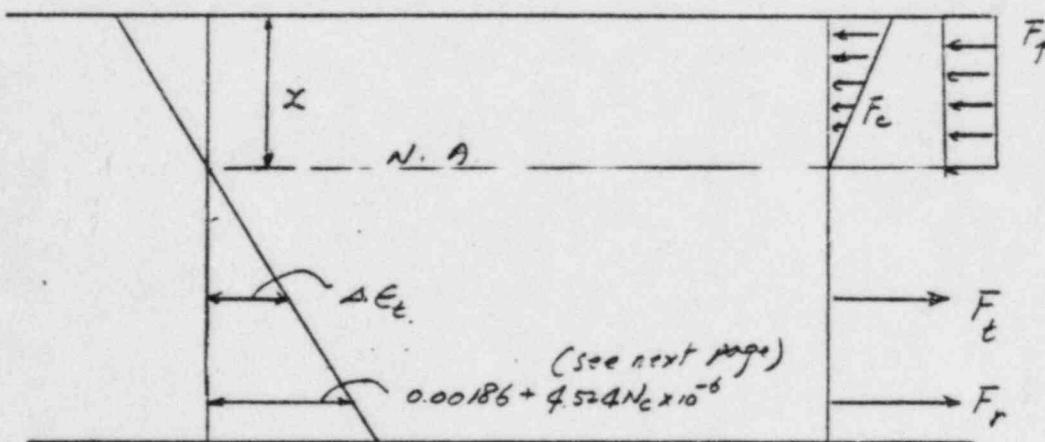


Fig. B-9

strain due  
to momentForces due  
to moment & prestress

## CALCULATION SHEET

## CALCULATIONS FOR

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1  
2  
3 Due to prestress:  
4

5 Tendon stress at end of life:  
6

$$7 \quad 0.7 f_s' (1 - 0.12)(0.885) = 0.575 f_s' = 130.8 \text{ ksi}$$

$$8 \quad \text{Loss Friction effect}$$

$$9 \quad (\text{Ref. 1, Table 5.6-4}), (\text{Ref. 3})$$

10  
11 Corresponding strain in tendon is.  
12

$$13 \quad 0.00482 \text{ in./in. (Fig. 4)}$$

14  
15 Prestress in concrete:  
16

17 Total tendon force at end of life:  
18

$$19 \quad F_{t,e} = \frac{2}{3} N_c' \times 0.35 \times 130.8 / 294$$

$$20 \quad = 2.477 N_c' \text{ kip/in. width}$$

21 Concrete stresses:  
22

$N_c'$  = actual number of  
circ. head tendons.

$$23 \quad \text{Top (or bottom head)}: \quad \frac{2.477 N_c'}{186} - \frac{(2.477 N_c')(118 - \frac{186}{2})}{6}$$

$$24 \quad = 0.0133 N_c' - 0.0107 N_c' = 0.0026 N_c' \text{ k/in.}^2$$

$$25 \quad \text{Bottom} = 0.0133 N_c' + 0.0107 N_c' = 0.024 N_c' \text{ k/in.}^2$$

26 Concrete strains:  
27

$$28 \quad E_c = 5.0 \times 10^6 \text{ psi. (Ref. 1, § 5.4.1)}$$

$$29 \quad E_{top} = 0.52 N_c'^{-0.4} \text{ in./in.}, \quad E_s = 4.8 N_c' \times 10^{-6} \text{ in./in.}$$

## CALCULATION SHEET

## CALCULATIONS FOR

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1

2 Concrete strains:

3

4

at rebar location:

5

6

$$\left[ 4.8 - \frac{4.8 - 0.52}{186} \cdot 12 \right] N_c \times 10^{-6} = 4.524 N_c \times 10^{-6}$$

7

8

9

Due to Moment:

10

11

12

Limit: Max. rebar stress = 0.9  $f_{sy}$  = 54 ksi.

13

14

Rebar strain at limit:

15

16

17

$$\frac{54}{29000} = 0.00186 \text{ in./in.}$$

18

19

Rebar strain due to moment:  $\epsilon_r = 0.00186 + 4.524 N_c \times 10^{-6}$ 

20

21

With  $N_c' = 34$  (Max.), additional  $\epsilon_r$  over 0.00186

22

23

is 0.000156 or about 8% of 0.00186.

24

25

For simplicity and conservatism, assume

26

27

$$\epsilon_r = 0.00186 \text{ in./in.}$$

28

29

This gives:

30

31

$$f_r = \text{re'stress} = 54 \text{ ksi}$$

32

33

$$F_r = 0.12 \times 54,000 = 6480 \text{ lbs/in.}$$

34

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# CALCULATION SHEET

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1  
2 Tendons:

3  
4  $\Delta \epsilon_t = \frac{118 - x}{174 - x} : 0.00186$

5  
6  $f_t = 130.8 + \Delta \epsilon_t \cdot 27,000 = 130.8 + \frac{118 - x}{174 - x} \cdot 50.28$   
7 (psi)

8  $F_t = \frac{\epsilon}{3} N_c' \times 0.35 \times f_t / 294$   
9  
10  $= 18.9 f_t N_c' \quad (lbs/in.)$   
11 (in<sup>4</sup> ksi)

12 Concrete:

13 Max. strain:  $\frac{x}{174 - x} \epsilon_r = \epsilon_c = \frac{x}{174 - x} 0.00186$

14 Max. stress.  $f_c = E_c \epsilon_c = 5 \times 10^6 \times \frac{x}{174 - x} 0.00186$   
15  
16  $= \frac{9300x}{174 - x} \quad (psi)$

17 Total force in concrete:

18  $F_c = \frac{1}{2} x \cdot f_c = \frac{4650x^2}{174 - x} \quad (lbs/in.)$

19 Effect of pressure inside penetration:

20  $F_p = 610x \quad (lbs/in.)$

## CALCULATION SHEET

CALCULATIONS FOR

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Equation of equilibrium:

$$F_c + F_p = F_t + F_r$$

$$\frac{4650x^2}{174-x} + 610x = 18.9N_c \left[ 130.8 + \frac{118-x}{174-x} \cdot 50.28 \right] + 6480$$

$$4650x^2 + 610x(174-x) = 18.9N'_c \cdot 130.8(174-x)$$

$$+ 18.9N'_c (118-x) 50.28$$

$$+ 6480(174-x)$$

$$(4650 - 610)x^2 + (610 \cdot 174 + 18.9 \cdot 130.8 N'_c + 18.9 \cdot 50.28 N'_c + 6480) x - (18.9 N'_c \cdot 130.8 \cdot 174 + 18.9 N'_c \cdot 118 \cdot 50.28 + 6480 \cdot 174) = 0$$

$$\frac{4040x^2 + (112620 + 3922N'_c)}{a} - \frac{(1127.520 + 5422.83N'_c)}{c} = 0$$

$$x = \frac{-b \pm \sqrt{b^2 + 4ac}}{2a}$$

$$\text{Capacity: } M = \frac{2}{3}x F_c + 305x^2 + F_t(118-x) + F_r(174-x)$$

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1

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## Table B-19 Moment Capacity

$N_c$	34	32	30	28	-
$a$					
$b$	-				
$c$	.				
$\bar{x}$ (in.)	46.80	45.52	44.18	42.78	
$f_{c,max}$ (psi)	3422	3295	3165	3031	
$F_c$ (lbs/in.)	80.070	74.988	69.914	64.854	
$f_t$ (ksi)	158.94	159.16	159.39	159.62	
$F_t$ (lbs/in.)	102138	96263	90375	84472	
$F_r$ (lbs/in.)	6480	6480	6480	6480	
$M$ ( $10^6$ lbs-in. per in.)	11.27	10.72	10.17	9.61	

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## CALCULATION SHEET

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Table B-19 Moment Capacity (Continued)

$N_c'$	24	22	20	16
$a$				
$b$				
$c$				
$N_A$				
$x$ (in.)	39.78	38.17	36.46	32.73
$f_{c,max}$ (psi)	2756	2613	2465	2155
$\bar{f}_c$ (lbs/in.)	54823	49877	44942	35261
$f_c$ (ksi)	160.10	160.35	160.61	161.15
$F_t$ (lbs/in.)	72622	66674	60710	48731
$F_r$ (lbs/in.)	6480	6480	6480	6480
$M$ ( $10^6$ lbs-in) per in.	8.49	7.92	7.34	6.17

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Table 8-19 Moment Capacity (Continued)

$N_c'$	14	12	10	8	6
a					
b					
c					
$N_A$ $\chi$ (in.)	30.66	28.43	26.00	23.33	20.55
$f_{c,max}$ (psi)	1989	1816	1634	1440	1232
$F_c$ (lbs/in.)	30495	25819	21239	16798	12533
$f_t$ (ksi)	161.44	161.74	162.06	162.39	162.75
$F_t$ (lbs/in.)	42716	36682	30628	24554	18456
$F_r$ (lbs/in.)	6480	6480	6480	6480	6480
$M$ ( $10^6$ lbs-in. per in.)	5.57	4.96	4.35	3.73	3.09

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1

2 Effect of pressure inside penetration. - a check.3  
4 If  $F_p = 6/10 x$  is neglected, the equation of  
5 equilibrium becomes.  
6  
7

8  
9  $F_c = F_t + F_r$

10  
11  $4650 x^2 + (6480 + 3422 N'_c) - (1127,520 + 542,282 N'_c)$   
12  
13  $= 0$

14  
15  $x = \frac{1}{2a} (-b + \sqrt{b^2 + 4ac})$

16  
17  $M = \frac{2}{3} x F_c + F_t (118 - x) + F_r (172 - x)$

18  
19  
20 Check  $N'_c = 34 \rightarrow x = 52.99$

21  
22  $M = 11.19 \times 10^6$

23  
24 If  $F_p$  is included.  $x = 46.80$ ,  $M = 11.27 \times 10^6$

25  
26  
27  $\frac{11.27 - 11.19}{11.19} = 0.01 \text{ off.}$

28  
29 Thus  $F_p$  effect is small.30  
31 Also, if  $N'_c = 0 \rightarrow M = 1.095 \times 10^6 \text{ vs } 1.10 \times 10^6$   
32  
33 ( $\text{No } F_p$ ) ( $\text{with } F_p$ )34  
35  
36

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B. 1.8 Required Number of Head Tendons

Table B-20.

$N_x$	1.0 RP			1.5 RP		
	Unit Yield Line Moment Req'd	$N_c'$ Req'd <sup>(1)</sup>	$N_c$ Req'd <sup>(1)</sup>	Unit Yield Line Moment Required	$N_c'$ Req'd <sup>(1)</sup>	$N_c$ Req'd <sup>(1)</sup>
0	16-in/in. $7.50 \times 10^6$	20.55	13.70	16-in/in. $16.34 \times 10^6$	(2)	-
6	$6.15 \times 10^6$	15.93	10.62	$14.72 \times 10^6$	(2)	-
12	$4.98 \times 10^6$	12.07	8.05	$13.38 \times 10^6$	(2)	-
18	$4.04 \times 10^6$	9.00	6.00	$11.63 \times 10^6$	34 <sup>(3)</sup>	22.67 <sup>(3)</sup>
24	$3.08 \times 10^6$	6.00	4.00	$10.11 \times 10^6$	29.79	19.86
Source	Tables B-15, B-16	Table 19 by interpolation	$N_c' \times \frac{2}{3}$	Tables B-17, B-18	Table 19 by interpolation	$N_c' \times \frac{2}{3}$

(1) Required to provide (See Fig. 1)

a moment capacity equal to the unit yield line moment.

(See Fig. 2)

(2) Not possible.

(3) Corresponds to  $N_x = 19.42$ . Max  $N_c'$  is 34.  $N_x = 18$  req'd  $N_c' > 34$ .

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1  
 2  
 3       The calculated effective number of circumferential  
 4       head tendons and the corresponding number of  
 5       crosshead tendons required are shown graphically  
 6       in Figs. 1 and 2, main text.  
 7  
 8  
 9  
 10

11       As a further conservatism, the curve for 1.5 RP  
 12       (Fig. 2) is truncated at  $N_c = 22$  since in each  
 13       head, the minimum  $N_c$  passing any section  
 14       is 22, out of 34 actual number of tendons  
 15       currently provided.  
 16  
 17  
 18  
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 20  
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## CALCULATION SHEET

## CALCULATIONS FOR

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8.2 Potential Failure Mode with YieldLine at Edge of Core Cavity

No.  $N_x$  requirements were determined from consideration of yield patterns O-A-B-H-C and O-A-D-G-E. (See Fig. B-3).

Potential yield line failure along the yield pattern O-A-B-H-I and O-A-D-G-I are checked to assure that any of these will not be critical.

(See Fig. B-3)

Four  $N_x$ ,  $N_c'$  combinations representing the extreme conditions in Figs. 1 and 2 are checked. They are

$$N_x = 0, \quad N_c' = 20 \quad (N_c = 13) \quad \} \text{Fig. 1}$$

$$N_x = 24, \quad N_c' = 6 \quad (N_c = 4)$$

$$N_x = 21, \quad N_c' = 33 \quad (N_c = 22)$$

$$N_x = 24, \quad N_c' = 30 \quad (N_c = 20)$$

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Bottom Rebars for Bottom Head (Ref. 7)

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5

Lager #1 : A11 #8, A432

6

7

$$LBH-1, 2, \dots, 9 = 102$$

8

9

$$LBH-10, 11, 12, 13, 14 = 72$$

$$\left. \begin{array}{l} 184 \times (0.79) = 145.36 \\ \text{in.} \end{array} \right\}$$

10

11

12

Layers #3, 5, 7: A11 #8, A432

13

14

$$LBH-18, 64$$

15

16

$$LBH-21 32$$

17

18

$$11' - 11" \times 9, 24$$

19

20

$$11' - 2" .. 36$$

21

22

$$12' - 9" .. 24$$

23

24

$$7' - 6" .. 12$$

$$\left. \begin{array}{l} 192 \times (0.79) = 151.68 \text{ in}^2 \\ \text{in}^2 \end{array} \right\}$$

25

26

Total in radial dir. = 297.04 in<sup>2</sup>.

27

28

$$\text{Or. } \frac{297.04}{2\pi \cdot 186} = 0.254 \text{ in}^2/\text{in liner}$$

29

30

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## CALCULATION SHEET

## CALCULATIONS FOR

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1                   2 Top Rebars for Bottom Head

3                   4 Diagonal Bars (Refs. 8. & 10) A431, #11

$$5 \quad 12 \times 16 \times (1.56) \times \cos 19^{\circ} 50' \times \cos 45^{\circ}$$

$$6 \quad \#11 \quad (\text{plan}) \quad (\text{Elevation})$$

$$7 \quad = 192 \times 1.56 \times \cos 19^{\circ} 50' \times \cos 45^{\circ}$$

$$8 \quad = 299.52 \text{ in}^2 \times \cos 19^{\circ} 50' \times \cos 45^{\circ}$$

$$9 \quad (19.85^{\circ})$$

$$10 \quad = 199.23 \text{ in}^2 \quad \text{total in radial direction}$$

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Radial Bars (Refs. 8 and 9) A431, #145

(Area)

$$\text{UBH-5, 5A, 6, 6A: } \delta \times 2.25 \times \cos 27.5^{\circ} \times \cos 9^{\circ} = 15.77$$

$$\text{UBH-7, 7A, 8, 8A: } \delta \times 2.25 \times \cos 27.5^{\circ} \times \cos 18^{\circ} = 15.18$$

$$\text{UBH-9, 9A, 10, 10A: } \delta \times 2.25 \times \cos 27.5^{\circ} \times \cos 25^{\circ} = 14.47$$

$$\text{UBH-11, 11A, 12, 12A: } \delta \times 2.25 \times \cos 27.5^{\circ} \times \cos 33^{\circ} = 13.39$$

$$\text{UBH-13, 17A, 18, 18A: } 16 \times 2.25 \times \cos 28^{\circ} \times \cos 9^{\circ} = 31.39$$

$$\text{UBH-19, 19A, 20, 20A: } 16 \times 2.25 \times \cos 28^{\circ} \times \cos 16^{\circ} = 30.23$$

$$\text{UBH-21, 21A, 22, 22A: } 16 \times 2.25 \times \cos 28^{\circ} \times \cos 25^{\circ} = 28.81$$

$$\text{UBH-23, 23A, 24, 24A: } 16 \times 2.25 \times \cos 28^{\circ} \times \cos 33^{\circ} = 26.66$$

$$\text{UBH-33, 33A, 34, 34A: } 96 \times 2.25 \times \cos 27.5^{\circ} \times \cos 27.5^{\circ} = 170.32$$

$$\text{Total in radial dir. = 396.23}$$

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2

3 Total in radial dir. = 199.23 + 386.23 = 585.46 in<sup>2</sup>

4

5

6

Or  $\frac{585.46}{2\pi \cdot 186} = 0.467 \text{ in}^2/\text{in.}$

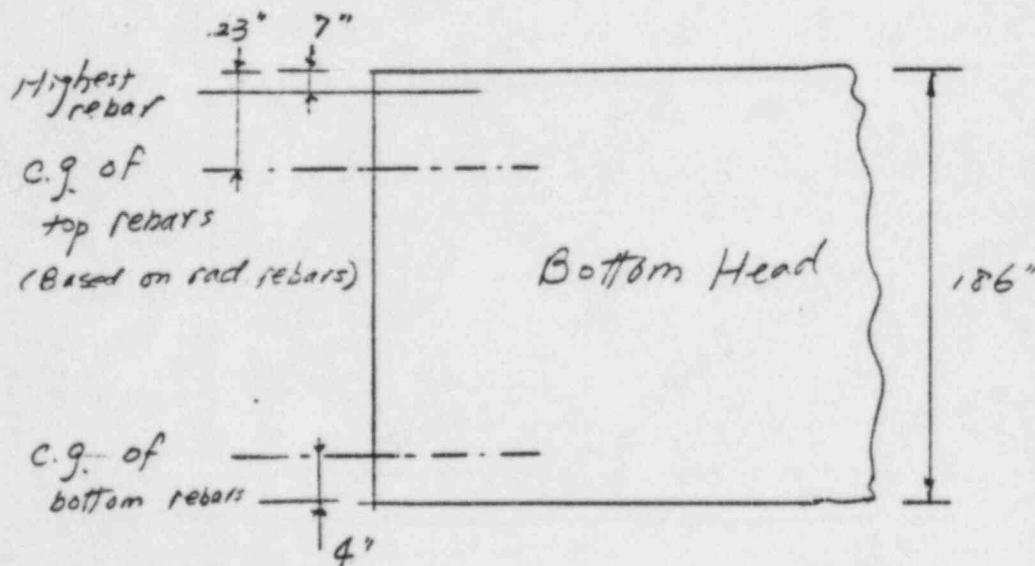
7

8

9

10 Location of Rebars

11 (Refs. 8 and 11)



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Radial prestress due to Crosshead Tendons

Referring to Ref. 10,

of 24 tendons, 12 are essentially radial at the edge of core cavity, and 12 are inclined about  $30^\circ$  w.r.t. the radial direction. Vertically, the largest inclination is  $43^\circ$ . Hence the prestress force is:

$$2 \cdot \frac{N_x}{2} (1 + \cos 30^\circ) \cos 43^\circ \times 0.35 \times f_x$$

(stress in X-head tendons.)

$$= 2 \times 5.698 f_x N_x \text{ kips}$$

(f<sub>x</sub> in ksi)

$$\text{or } \frac{2 \times 5.698 f_x N_x}{2 \pi \cdot 186} = 9.75 f_x N_x \text{ lbs/in.}$$

(f<sub>x</sub> in ksi.)

From Ref. 11, the location of crosshead tendons at the edge of core cavity is, about 8 ft from the top of bottom head

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Moment Capacity Along a Radial Yield Line

Same as before (See § B.1.7) except

$$\text{use } F_t = 18.9 (120.8) N_c = 2472 N_c^*$$

$$F_c + F_p = F_t + F_r$$

$$\frac{4650 x^2}{174-x} + 610 x = 2472 N_c + 6480$$

$$4650 x^2 + 610x(174-x) = (2472 N_c + 6480)(174-x)$$

$$4040 x^2 + (610 \times 174 + 2472 N_c + 6480)x$$

$$- 174(2472 N_c + 6480) = 0$$

$$\frac{4040 x^2}{a} + \frac{(112620 + 2472 N_c)x}{b} - \frac{(1127520 + 430128)}{c} = 0$$

$$x = \frac{-b + \sqrt{b^2 + 4ac}}{2a}$$

$$M = \frac{2}{3} x F_c + 3052^2 + F_t (118-x) + F_r (174-x)$$

\* The increase in prestress due to further extension of tendons is ignored.

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## Table 8-21 Moment Capacity

3

## Along a Radial Yield Line

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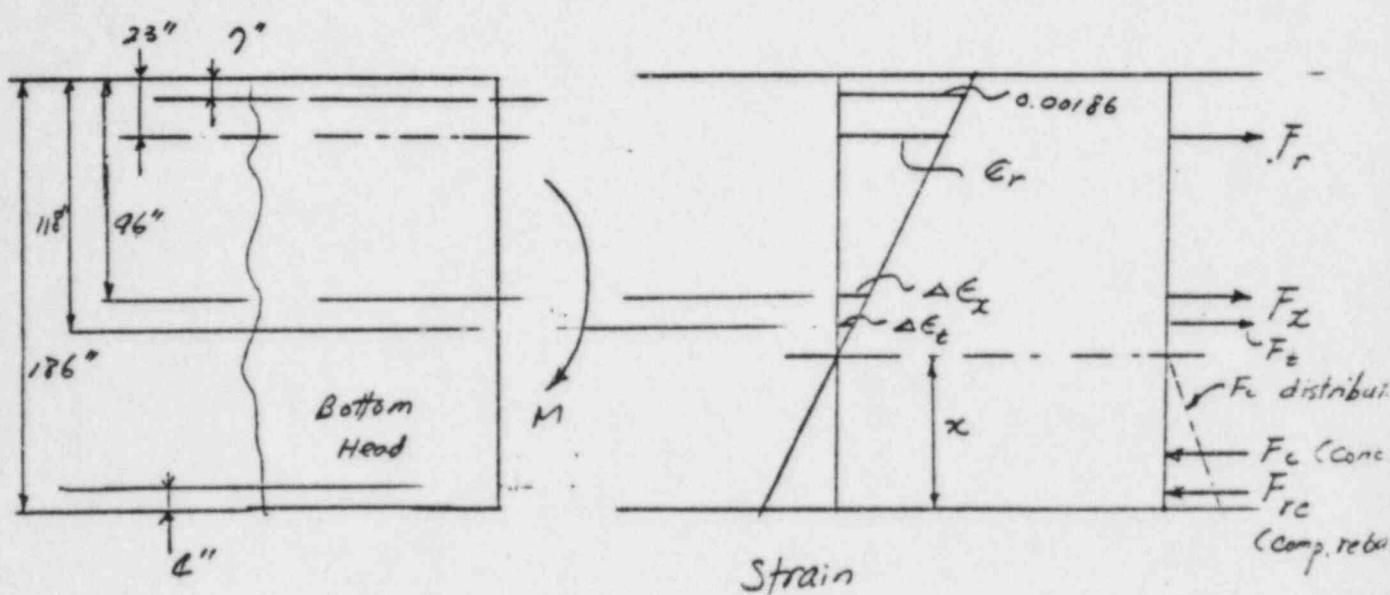
36

$N_c$	20	6	33	30
a	4040	4040	4040	4040
b	162060	127182	194196	186780
c	9,730,080	3,708,288	15,321,744	14,031,360
$\bar{x}$ (in.)	32.96	18.40	42.07	40.19
$f_{c,max}$ (psi)	2173	1100	2966	2793
$F_c$ (lbs/in.)	35,817	10118	62386	56128
$f_t$ (ksi)	130.8	130.8	130.8	130.8
$F_t$ (lbs/in.)	29440	14832	81576	74160
$F_r$ (lbs/in.)	6480	6480	6480	6480
$M$ ( $10^6$ lbs-in) per in.	6.24	2.71	9.34	8.63

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due to moment  
(≈ Strain at  
limit )

Fig. B-10

$$\text{Max. rebar strain} = \frac{54}{29000} = 0.00186$$

$$\text{Ave. rebar strain} = \frac{186-23-x}{186-7-x} \cdot 0.00186 = \frac{163-x}{179-x} \cdot 0.00186$$

Total rebar force (tensile):

$$F_r = 0.467 \times \frac{163-x}{179-x} (0.00186) (29,000,000)$$

(Area/in.)

$$= 25190 \times \frac{163-x}{179-x} \quad \text{lbs/in.}$$

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1

2

3 Total rebar force (compressive):  
4

5 Strain:  $0.00186 \frac{x-4}{179-x}$   
6

7  $F_{rc} = 0.00186 \frac{x-4}{179-x} \cdot 29,000,000 \times 0.254$   
8 (Area/in.)  
9  
10  $= 13700 \frac{x-4}{179-x} \text{ lbs/in.}$

11

12 Grosshead tendons:  
13

14  $\Delta \epsilon_x = \frac{186-96-x}{179-x} \cdot 0.00186 = 0.00186 \frac{90-x}{179-x}$   
15

16 Stress at end of life:  
17

18  $0.7 f'_s (1 - 0.12)(0.9) = 0.554 f'_s = 133 \text{ ksi}$   
19 loss friction  
20

21  $f_x = \text{Stress in X-head tendons}$   
22

23  $= 133 + \Delta \epsilon_x \cdot 27,000 = 133 + \frac{90-x}{179-x} \times 50.28 \text{ (ksi)}$   
24

25  $F_x = 9.75 f_x \cdot N_x \text{ (lbs/in.)}$   
26  
27 (in ksi)  
28

29

30

Concrete:

31  $\text{Max. strain} - \epsilon_c = 0.00186 \frac{x}{179-x}$   
32

33  $f_c(\text{max}) = E_c f_c = 5 \times 10^6 \times 0.00186 \frac{x}{179-x} = \frac{9300x}{179-x} \text{ (psi)}$   
34

35  $F_c = \frac{1}{2} x f_c = \frac{4650x^2}{179-x}$   
36

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$$F_t = \frac{2}{3} N_c' (0.35) \cdot f_t / 22.17 \times 12 = 20.9 f_t N_c' (\text{lb/in.})$$

E Ave. radius of cir. tendon

Use  $f_t = 130.8 \text{ ksi}$ . (Inc. in  $f_t$  due to  $\Delta \epsilon_4$  ignored.)

$$F_t = 20.9 \times 120.8 \text{ N}_c = 2734 \text{ N}_c$$

### Equilibrium:

$$F_r + F_x + F_c = F_c + F_{rc}$$

$$25190 \cdot \frac{163-x}{179-x} + 9.75(133 + \frac{90-x}{179-x} \cdot 50.28) N_x$$

$$+ 2734 \quad N_c = \frac{4650x^2}{179-x} + 13700 \quad \frac{x-4}{179-x}$$

$$25190(163-x) + 9.75(133)(179-x)N_x + 9.75(90-x)50.28\lambda$$

$$+ 2734 N_c (179-x) = 4650 x^2 + 13700 (x-4)$$

$$4650 x^2 + (13700 + 25790 + 9.75 \times 133 N_x + 9.75 - 50.28 N_x + 2732 N_y)$$

$$= (13700 \times 4 + 25190 \times 163 + 9.75 \times 133 \times 179) N_L + 9.75 \times 90 \times 50.28 N_L$$

$$+ 2734 \times 179 N_C) = 0$$

$$\frac{4650}{a} x^2 + \frac{(38890 + 1787 N_x + 2734 N_c)}{b} x$$

$$-\frac{(2,160,770 + 276,239 N_2 + 289,386 H)}{C} = 0$$

$$x = \frac{1}{2a} (-b + \sqrt{b^2 + 4ac})$$

$$M = F_r(163-x) + F_x(90-x) + F_t(68-x) + F_c\left(\frac{4}{3}x\right) + \bar{F}_n(x)$$

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## CALCULATIONS FOR

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Table B-22 Moment Capacities Along a Hoop Yield Line  
 (At  $R = 186$  in.)

$N_x$	0	24	21	28
$N'_c$	20	6	33	30
$x$ (in.)	45.62	44.79	59.13	58.47
$F_r$ (lbs/in.)	22168	22187	21828	21826
$f_y$ (ksi)	-	149.94	145.95	146.15
$F_t$ (lbs/in.)	0	35085	29883	34200
$F_c$ (ksi)	130.8	130.8	130.8	130.8
$F_{tc}$ (lbs/in.)	54680	16404	90222	82020
$f_c$ (psi)	3181	3104	4588	4512
$F_c$ (lbs/in.)	72556	69507	135631	131894
$F_{rc}$ (lbs/in.)	4275	4164	6301	6191
$M$ (lbs-in./in.)	6.21	6.83	9.68	9.62

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE B71 OF
PREPARED BY T. T. Lee	DATE 11/19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

1

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Yield Line Pattern OADG1

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Referring to Fig. B-6, the forces acting on the slab section bounded by OA'DGIG'D'A consist of the effective pressure on the surface, vertical force ( $F_{L-L}$ ) along GIG', shearing forces along OADG and OA'D'G', moment along the two radial yield lines (OADG and OA'D'G'), and moment along the hoop yield line GIG'. For simplicity ignore the shear along OADG, & OA'D'G'.

From equilibrium, the moment of the effective pressure and  $F_{L-L}$  about any horizontal axis must be balanced by the moments along all yield lines. For convenience, use E-E' as the axis to compute the moment due to pressure and  $F_{L-L}$ .

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE 872 OF
PREPARED BY T. T. Lee	DATE 11/19/82	REF. DOCUMENTS:	
CHECKED BY	DATE		

1

2

3 Check 1 RP,  $N_x = 0$ ,  $N_c' = 20$  Case

4

5

From Table B-16,

6

7

(1)  $M_{E-E}$  due to pressure  $= 1.862 \times 10^9$  lb-in.

8

9

(2) " " "  $F_{L-L}$   $= -0.946 \times 10^9$  lb-in.

10

11

Sum  $= 0.916 \times 10^9$

12

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$= 916 \times 10^6$  lb-in.

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Resisting Moment

$$(3) A \text{ along } OADG \times OAD'G' = 2 \times \underline{6.24 \times 10^6} \times \underline{(9+1+9)} \quad (\text{Table B-21})$$

$$= 237 \times 10^6 \quad (\text{See Fig. B-7})$$

$$(4) A \text{ along } GIG' \quad (\text{Table B-22})$$

$$= 2 \times \underline{6.21 \times 10^6} \times \underline{186 \times 5 \times 10^6} \quad (\text{See Fig. B-6})$$

$$= 712 \times 10^6$$

Sum  $= 951 \times 10^6$  lb-in.

$> 916 \times 10^6$  lb-in.

This yield line failure mode does not control

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE 873 EF
PREPARED BY T. T. Lee	DATE 11/19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

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Check 1RP.  $N_x = 24$ ,  $N_c' = 6$  Case

$$M_{E-E'} \text{ due to pressure} = 1.214 \times 10^9$$

$$" " + F_{L-L} = -0.617 \times 10^9$$

---


$$\text{Sum} = 0.597 \times 10^9$$

$$= 597 \times 10^6 \text{ lb-in.}$$

Resisting Moment.

$$\text{Along OADG + OAD'G'} = 2 \times 2.71 \times 10^6 \times 19 = 103 \times 10^6$$

$$\text{Along GIG'} = 2 \times 6.83 \times 10^6 \times 186 \times \sin 18^\circ$$

$$= 785 \times 10^6$$

---


$$\text{Sum} = 588 \times 10^6 \text{ lb-in.}$$

$$> 597 \times 10^6 \text{ lb-in.}$$

O.K.

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO.	907738	N/C	PAGE 87 OF
PREPARED BY T. T. Lee	DATE 11/19/82	REF. DOCUMENTS:			
CHECKED BY	DATE				

1

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4

Check 1.5 RP,  $N_x = 21$ ,  $N_c' = 33$  Case

5

6

From Table B-18,

7

8

$$M_{E-E'} \text{ due to pressure} = 2.224 \times 10^9 \text{ (by interpolation)}$$

9

10

$$\text{" " " } F_{L-L} = -1.130 \times 10^9 \text{ (by interpolation)}$$

11

12

$$\text{Sum} = 1.094 \times 10^9 = 1094 \times 10^6 \text{ lb-in}$$

13

14

Resisting moment

15

16

$$\text{Along OADG, OA'D'G'} = 2 \times 9.34 \times 10^6 \times 19 = 355 \times 10^6$$

17

18

$$\text{Along GIG'} = 2 \times 9.68 \times 10^6 \times 188 \times \sin 18^\circ = 1113 \times 10^6$$

19

20

21

$$\text{Sum} = 1268 \times 10^6 > 1094 \times 10^6 \text{ lb-in}$$

O.K.Check 1.5 RP,  $N_x = 24$ ,  $N_c' = 30$  Case

22

23

$$M_{E-E'} \text{ due to pressure} = 2.142 \times 10^9$$

24

25

$$\text{" " " } F_{L-L} = -1.088 \times 10^9$$

26

27

$$\text{Sum} = 1.054 \times 10^9 = 1054 \times 10^6 \text{ lb-in}$$

28

Resisting moment

29

30

$$\text{Along OADG, OA'D'G'} = 2 \times 8.63 \times 10^6 \times 19 = 328 \times 10^6$$

31

32

$$\text{Along GIG'} = 2 \times 9.62 \times 10^6 \times 188 \times \sin 18^\circ$$

33

34

$$= 1106 \times 10^6$$

35

36

$$\text{Sum} = 1234 \times 10^6 > 1054 \times 10^6 \text{ lb-in}$$

O.K.

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO.	907738	N/C	PAGE 8756r
PREPARED BY T. T. Lee	DATE 11/19/84	REF. DOCUMENTS:			
CHECKED BY	DATE				

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Yield Line Pattern OABHI

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5

Refer to Fig. B-6.

6

7

Check 1.0 RP,  $N_x = 0$ ,  $N_c' = 20$  case

8

9

10

From Table B-15.

11

12

$$M_{c-c} \text{ due to pressure} = -599 \times 10^9$$

13

14

$$\text{---} \quad \text{---} \quad F_{L-L} = -1.648 \times 10^9$$

15

16

$$\text{Sum} = 0.951 \times 10^9$$

17

18

$$= 951 \times 10^6 \text{ lb-in}$$

19

Resisting moment

20

21

$$\text{along OABH, OA'B'H'} = 2 \times 6.24 \times 10^6 \times (9+13+9) = 3072 \times 10^6$$

22

23

$$\text{along HIH'} = 2 \times 6.21 \times 10^6 \times 186 \times \sin 41.5^\circ = 1531 \times 10^6$$

24

25

$$\text{Sum} = 1918 \times 10^6 \text{ lb-in} > 951 \times 10^6$$

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O.K.

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO.	907738 N/C	PAGE 8760F
PREPARED BY T. T. Lee	DATE "/19/84	REF. DOCUMENTS:		
CHECKED BY	DATE			

1

2 Check 1.0 RP,  $N_x = 24$ ,  $N_c' = 6$  Case

3

4  $M_{c-c}$  due to pressure =  $2.374 \times 10^9$

5

6 " " "  $F_{c-a} = -1.505 \times 10^9$

7

8  $\sum_m = 0.869 \times 10^9 = 869 \times 10^6$  lb-in

9

10 Resisting moment

11

12 along OABH, OA'B'H' =  $2 \times 2.71 \times 10^6 \times 31 = 168 \times 10^6$

13

14 along HIH' =  $2 \times 6.83 \times 10^6 \times 186 \times \sin 41.5^\circ = 1684 \times 10^6$

15

16  $\sum_m = 1852 \times 10^6 > 869 \times 10^6$  lb-in

17

OK

18

19 Check 1.5 RP,  $N_x = 21$ ,  $N_c' = 33$  Case

20

21  $M_{c-c}$  due to pressure =  $3.104 \times 10^9$

22

23 " " "  $F_{c-a} = -1.968 \times 10^9$

24

25  $\sum_m = 1.136 \times 10^9 = 1136 \times 10^6$  lb-in

26

27 Resisting moment:

28

29 along OABH, OA'B'H' =  $2 \times 9.34 \times 10^6 \times 31 = 579 \times 10^6$  lb-in

30

31 along HIH' =  $2 \times 9.68 \times 10^6 \times 186 \times \sin 41.5^\circ = 2386 \times 10^6$

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33  $\sum_m = 2965 \times 10^6 > 1136 \times 10^6$  lb-in

34

OK

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## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE E77C
PREPARED BY T. T. Lee	DATE 1/19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

Check 1.5 R.P.  $N_x = 24$ ,  $N_c' = 30$  Case

$$M_{c-cr} \text{ due to pressure} = -2.990 \times 10^9$$

$$\text{ " " } F_{c-cr} = -1.875 \times 10^9$$

$$\text{Sum} = 1.095 \times 10^9 = 1095 \times 10^6 \text{ lb-in}$$

Resisting moment:

$$\text{along OABH, OA'B'H'} = 2 \times 863 \times 10^6 \times 31 = 535 \times 10^6$$

$$\text{along HIH'} = 2 \times 9.62 \times 10^6 \times 186 \times \sin 41.5^\circ = 2371 \times 10^6$$

$$\text{Sum} = 2906 \times 10^6 > 1095 \times 10^6 \text{ lb-in}$$

O.K.

Conclusion (§B.1.9)

Yield line failure along OABH or OADGI will not develop, if  $N_x$ ,  $N_c$  shown in Table E-20 (or Figs 1 and 2) are used. Hence Table E-20, or Figs 1 and 2 remain valid

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE 878 OF
PREPARED BY T. T. Lee	DATE // 19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

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B. 3

Punching Shear Mode of Failure

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(Bottom Head)

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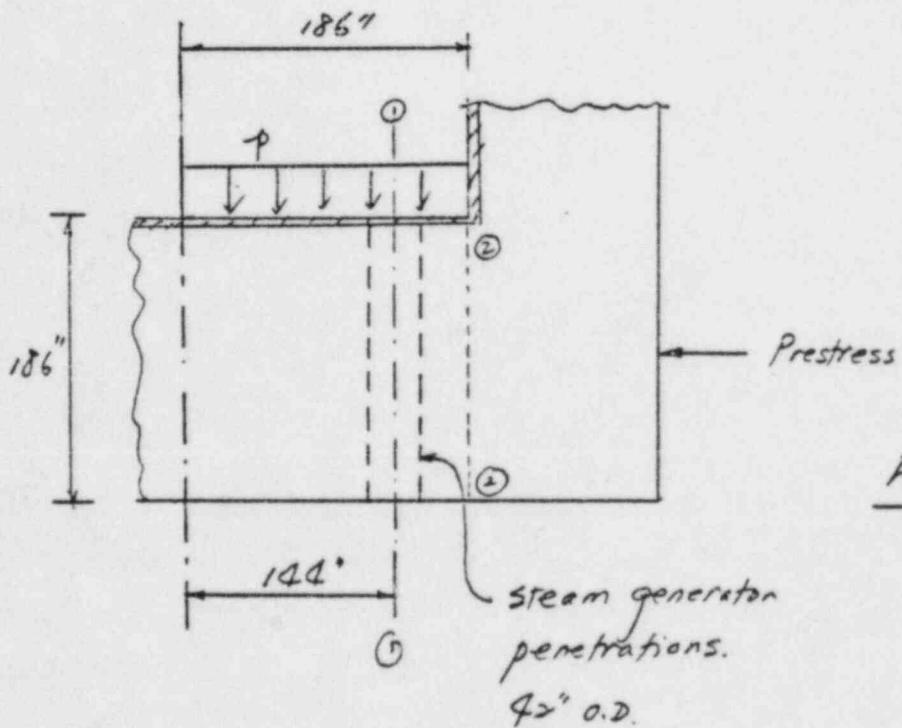


Fig. B-11

Section susceptible to punching shear failure

is the one connecting the S.G. penetrations,

Fig. B-11, ①-①. See Fig. B-3 also.

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE 279 OF
PREPARED BY T. T. Lee	DATE 11/19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

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The punching shearing stress is

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Ref 4 gives:

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$$\frac{p \cdot \pi (144)^2}{[2\pi(144) - 12(42)] \cdot 186} = 0.874 p = f_v$$

$$f = \frac{d}{D} (0.874 f'_c + 0.1676 f_c)$$

where  $p$  = ultimate pressure (psi)

$d$  = slab thickness = 186 in.

$D$  = pressurized diameter in inches = 278 in

$f'_c$  = concrete strength = 6000 psi

$f_c$  = prestress (psi)

From this, the shear allowable is:

$$\frac{p \frac{\pi D^2}{4}}{\pi d \cdot d} = \frac{p D}{4d} = f_{v, \text{allowable}}$$

## CALCULATION SHEET

## CALCULATIONS FOR

EQUIP. NO.	PROJ.	CALC. NO. 907738 N/C	PAGE 880 EF
PREPARED BY T. T. Lee	DATE 11/19/84	REF. DOCUMENTS:	
CHECKED BY	DATE		

$$\begin{aligned}
 f_{v, \text{allowable}} &= \frac{1}{4} (0.842 f_c' + 0.1676 f_t) \\
 &= 0.210 f_c' + 0.0419 f_t \\
 &= 1260 + 0.0419 f_t
 \end{aligned}$$

For  $p = 845 \text{ psig}$  .  $f_v = 738 \text{ psi}$   
 $(1.0 \text{ RP})$

$p = 1268 \text{ psig}$ ,  $f_v = 1108 \text{ psi}$   
 $(1.5 \text{ RP})$

Both are smaller than 1260 psi, the  
allowable ignoring the constraining effect  
of the radial prestress.

Check Section ②-② (Fig. B-11) :

$$\begin{aligned}
 \text{Allowable } p &= \frac{186}{2 \times 186} (0.842 f_c' + 0.1676 f_t) \\
 &= 2526 + 0.0838 f_t
 \end{aligned}$$

Actual  $p = 845$  and  $1268 \text{ psig} < 2526 \text{ psi}$

Hence, punching shear is not a problem

DD 907738/N/C

APPENDIX C

CALCULATION REVIEW REPORT

## CALCULATION REVIEW REPORT

## TITLE:

*FSV - Tendon Requirements Based on Safety Consideration*APPROVAL LEVEL 3  
QAL LEVEL 1

DISCIPLINE	SYSTEM	DOC. TYPE	PROJECT	DOCUMENT NO.	ISSUE NO/LTR.
S	II	CFL	1900	907738	N/C

## INDEPENDENT REVIEWER:

NAME Arnold A. SchwartzORGANIZATION Structural Design and AnalysisREVIEWER SELECTION APPROVAL: BR MGR Per. Ltr. DATE 10.29.84

## REVIEW METHOD:

ARITHMETIC CHECK

LOGIC CHECK

ALTERNATE METHOD USED

SPOT CHECK PERFORMED

COMPUTER PROGRAM USED

YES	NO	ERROR DETECTED
	✓	
✓		No
	✓	
✓		No
	✓	

## REMARKS: (ATTACH LIST OF DOCUMENTS USED IN REVIEW)

*Calculation 907738**FSV - Structural Dwg's for PCRV*

CALCULATIONS FOUND TO BE VALID AND CONCLUSIONS TO BE CORRECT:

INDEPENDENT REVIEWER

*Arnold A. Schwartz*

SIGNATURE

DATE

11/30/84