WISCONSIN ELECTRIC POWER COMPANY POINT BEACH NUCLEAR PLANT UNIT 2 CYCLE 11 STARTUP JANUARY, 1985

> BY R. L. HARRIS P. N. KURTZ



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PREFACE

This report is intended to document in a concise format the results of the physics testing program and unit systems response during the startup of Unit 2 following Refueling 10. The organization of the report follows that utilized previously in startup reports.

The core loading pattern was determined by Westinghouse, the vendor for the nuclear steam supply system. WCAP 10583, Revision 1 "The Nuclear Design - Core Management of the Point Beach Unit 2 Nuclear Reactor Cycle 11," tabulates various parameters predicted by computer codes. All references in this report to design values pertain to WCAP 10583. Actual end of Cycle 10 burnup was 13,677 MWD/MTU. The published WCAP parameters were based on actual Cycle 10 EOL burnup. Cycle 10 was ended on September 28, 1984 with a peak assembly burnup of 42,730 MWD/MTU and average assembly burnup of 24,108 MWD/MTU. Electrical power was first generated during Cycle 11 on November 20, 1984.

This report is intended primarily for the use of Wisconsin Electric Power Company personnel as a readily accessible, complete compilation of reduced data.

Copies of this report were submitted to the NRC to comply with Technical Specification 15.6.9.1.A.1.c and 15.6.9.1.A.2. A region of Westinghouse optimized fuel assemblies (OFA's) was loaded for the first time at PBNP in Unit 2 Cycle 11. The fuel design changes for OFA's were significant enough to be classified as constituting a different fuel design.

REFUELING

Section 1.0

Section 1.1 Core Unload

The core was completely unloaded to facilitate incore thimble changeout and reactor vessel component inspections. The first fuel assembly was unloaded on October 13, 1984 at 1842 hours. Using two 10-hour shifts per day, the unload was completed without any changes to the sequence on October 15, 1984, at 1752 hours.

All fuel was stored in the north spent fuel pit. Spent fuel receipt was suspended between core unload and core reload with one spent fuel assembly (D14) not put in the spent fuel pit.

There were no insert changes made during core unload.

One fuel assembly (M55) sustained grid damage when being placed in storage location SM-27. It was replaced with fuel assembly NO2 for the core reload.

Section 1.2 Insert Changes

- Eight RCCA's were replaced because of wear found during visual inspections performed in 1983. All control rod transfers were made without incident.
- Several depleted burnable poison (BP) assemblies were removed from or transferred between reload fuel assemblies with no incident.
- 3. Three new BP assemblies were transferred between new fuel assemblies. A fourth BP assembly could not be transferred because it repeatedly fell from the tool's gripper mechanism when lifted from a new fuel assembly. One of the new BP assemblies that were successfully transferred was then partially withdrawn by the tool in front of the periscope. It was discovered that the BP assembly crossbar was wedged in the gripper mechanism below the latching fingers. Apparently the three transfers were made with the BP assemblies held in the tool by friction.

The fourth BP assembly was transferred to the new fuel vault so that it could be inspected at a later date. It was replaced with a new BP assembly left over from Unit 1. 4. One plug device was damaged when a top nozzle spring clamp with orientation hole broke off and wedged between the plug device tool and top nozzle of fuel assembly K77. Several spare plug devices were available for replacement because of the changeover to optimized fuel requiring new redesigned plug devices. A spare plug device was put in fuel assembly K68 (replacement for K77).

All other plug device changes were made without incident.

Section 1.3 Fuel Assembly Inspections

OFA demonstration assemblies ZD1, ZD2, and ZD4 were inspected by Westinghouse. These assemblies have removable rods and have had 3 cycles of burnup. The inspection program included general visual examinations of the fuel assemblies and high magnification visual examination of several individual fuel rods. No abnormalities were found.

Section 1.4 Core Reload

Changes were made to the original core loading plan for Cycle 10 because of damage to the following fuel assemblies:

- M55 Replaced with NO2 after sustaining grid damage from spent fuel pit storage rack at location SM-27.
- K77 Replaced with K68 after the top nozzle spring clamp with orientation hole broke off.

As a result of the above replacements, changes were made to the original core loading sequence as described in Table 1-1.

Numerous changes had to be made to the core loading sequence because assemblies were bowed. This problem is expected to occur during full core reloads.

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TABLE 1-1

CHANGES TO CORE LOADING PLAN

	or	iginal	Fi	nal
Core Location	F/A	Insert	F/A	Insert
E-3	M55	RCCA	M77	RCCA
D-4	M77	RCCA	N02	RCCA
G-1	ZD2	ZPD	K68	PD
G-13	K77	8P50	ZD2	ZPD
D-3	N81	2P105Z	N81	PDZ

Section 1.5 Core Design

1. Optimized Fuel Assemblies

A region of 32 new optimized fuel assemblies were used for the first time at PBNP in Cycle 11. Their distribution in core is typical of the low leakage concept in which new fuel assemblies are loaded between the center area and extreme periphery of the core.

Three demonstration optimized fuel assemblies with removable fuel rods were loaded for a fourth cycle of operation at peripheral locations A-7, G-13, and M-7. These assemblies were found to be in good condition when inspected prior to core load.

The optimized fuel assembly employs a slightly reduced fuel rod clad OD (0.400 inch) compared to the standard fuel rod clad OD (0.422 inch) while retaining the same fuel rod pitch. This increases the water to uranium ratio which improves neutron moderation and efficiency eventually lowering fuel cycle costs. The fuel pellets are enriched to 3.4% in U-235.

Another feature of the optimized fuel assembly design is the use of zircaloy spacer grids for all but the top and bottom spacer grids. The top and bottom spacer grids are Inconel, the same material used in standard fuel assembly spacer grids.

Slight reductions in the guide thimble and instrument thimble diameters were also made. Standard control rods and burnable poison rods are compatible with optimized fuel assemblies. Standard plug devices, all having thicker plugging rods are not compatible however, and new plug devices were provided for use in optimized fuel assemblies.

2. Inserts

New burnable poison assemblies were provided in Cycle 11 to control radial power distribution. Seven 2P and eight 14P asymmetric burnable poison assemblies were loaded in optimized fuel assemblies. Four 4P burnable poison assemblies were loaded in once-burned fuel near the core's center.

Eight control rods were replaced in a continuing program leading eventually to total replacement. A total of 10 original control rods have been replaced since the program started in 1983.

The two secondary sources were returned to their normal locations at G-2 and G-12.

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3. Fuel Loading

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Table 1-2 lists the uranium weight by region. Figure 1-1 shows the final core load pattern, Figure 1-2 BOL SNM data, and Figure 1-3 BOL burnup data.

TABLE 1-2

URANIUM LOADING

	Number of	U Weigh	t (MTU)	Current
Region	Assemblies	Original	Current	Enrichment (%U235)
9A	. 1	0.40	0.38	0.74
10	5	2.00	1.92	1.10
10A	3	1.06	1.01	0.75
11	23	11.24	10.86	1.23
13A	4	1.60	1.55	1.36
13B	1	0.40	0.39	1.65
12	7	2.81	2.76	2.05
12A	40	16.13	15.80	1.88
13*	32	11.40	11.40	3.40
	TOTAL	47.04	46.07	2.02

* New Assemblies

FIGURE 1-1

CORE LOADING

PENP SHM DATA - UNIT 2 CYCLE 11 - START OF CYCLE AS OF 10/26/84

A

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11/14/84

1 2 3 4 5 6 7 8 9 16 11 12 13

LAO ZD1 LA2 LHOAFQ LHO3PK LHOAFL PB SPB PB

L52 N53 N60 N65 N63 N73 L77 LNOAFY LNOE3S LNOE3K LNCBVD LNOE2X LNOE38 LNOAFC PD QPD R 64 PD R 84 QPD PD

L72 N74 N52 N07 L63 L58 N58 N52 L55 LNOAF6 LNOE37 LNOAV9 LNO9YB LNOAFH LNOAFT LNOAVE LNOE3T LNOAFH PB 2P 107 R 56 PB R 114 PB R 65 2P 100 PB

L66 NG1 NG2 N79 N53 K59 N68 M97 N82 N69 L67 LNOAFH LNOE38 LNO9YJ LNOBVE LNOE38 LNO3NU LNOE31 LNOBUS LNOE33 LNOAFK P3 QP3 R 76 P3 14P102 P3 14P100 P3 R 74 2P 102 P3

N45 N77 N85 N44 N92 N88 N91 N81 N87 N57 N44 LN0532 LN08VA LN08UR LN08VF LN08UU LN0BUH LN08UE LN08UZ LN0BUH LN0AVF LN0534 UPB R 83 PB R 57 PB PB R 48 PB R 48 PB R 72 UPB

LS9 NS9 LS4 N42 N47 N76 N89 N68 N43 N51 N08 N72 L61 F LNOAFS LNOESL LNOAFX LNOESH LNOEV9 LNOBUS LNOBUF LNOBUT LNOBVN LNOESU LNO9YC LNOESS LNOAFJ P8 R 55 P8 14P105 P8 R 80 4P 102 R 69 P8 14P104 P8 R 107 PB

 K48
 H41
 L73
 K31
 H80
 H70
 J76
 H98
 H90
 K41
 L71
 H95
 ZB2

 8
 LH03PE
 LH08PK
 LH04FF
 LH03P0
 LH05V6
 LH07U7
 LH08UE
 LH03P1
 LH03P1</

LS1 NS8 NOS NS7 N66 N69 N96 N83 N94 N79 LS3 N77 L70 H LNOAGO LNOE3H LNOYYF LNOE3P LNOBUB LNOBUS LNOBUK LNOBUG LNOE3E LNOAFZ LNOE3C LNOAF9 PC R 61 PB 14P103 PB R 79 4P 103 R 63 PB 14P107 PB R 110 PB

N71 N56 N66 N74 N59 N75 N60 N73 N84 N51 N82 LNOEZY LNGAVE LNGBUL LNGBV4 LNGBUJ LNGBV6 LNGBUP LNGBV2 LNGBUT LNGAVA LNGE3F OP9 R 75 P5 R 62 P5 P5 R 81 P5 R 64 OP5

L75 N70 N93 N62 N67 K70 N76 N71 N78 N54 L74 LHOAFB LHOE36 LHOBUS LHOBUJ LHOE27 LHOBPC LHOE39 LHOBU1 LHOBUC LHOE3N LHOAFE PS 2P 103 R 73 PB 14P106 PB 14P101 PB R 82 2P 104 PB

LAP NEO NEA LAA LET NOA NEE NEA LAE LHOAFE LHOEED LHOAVE LHOAFE LHOAFE LHOAVE LHOEER LHOAFP PE 2P 101 R 54 PE R 114 PE R 34 2P 106 PE

> L78 NG1 N75 N72 NG4 N78 L68 LNOAFA LNOEJJ LNOEJA LNOBYO LNOEJO LNOEJB LNOAF7 PB OPB R 111 PB R 109 OPB PB

> > LSA ZB4 L76 LHOAFV LHO3PN LHOAFB PB SPD PB

> > > 6

FIGURE 1-2

BOL SNM DATA

	P8:1P 3:	M DATA	T UNIT	S CACFE	. 11 •	START	of cycl	E AS OF	10/26/	64			11/14/84
	1	a	3	4	5	٠	7	8	9	10	11	12	13
						150 4584 2434	201 2509 2020	L62 4517 2444					
8				L52 4574 2447	N53 12216 0	N60 12021 0	M65 9227 1414	12230 0	N73 12139 0	L77 4528 2448			
c			L72 4987 2371	N74 12038	M52 8084 1744	N07 5261 2232	L63 4846 2385	1255 2351	M58 8126 1726	NS2 12066	4981 2375		
ò		100 2442	N81 12116	N02 6446 2019	M79 7290 1949	N55 12040	×59 4543 2493	N68 12981	N97 7346 1959	M82 7326 1941	N69 11998 0	4527 2434	
٤		N65 11983	N77 7402 1923	M65 7335 1954	M64 6737 2078	2049 2005	M88 7337 1938	H91 6743 2986	M81 6713 2083	H87 7254 1951	NS7 8070 1720	12055 0	
,	L39 4556 2439	N59 12034	L54 5185 2353	12028	M67 6665 2097	M76 6758 2070	N69 8967 1521	M68 6779 2973	M63 6665 2087	N51 12028 0	N08 5318 2240	N72 12039 0	L61 4605 2437
6	K68 3121 2644	M61 9297 1414	L73 4872 2396	KS1 4506 2492	N80 7419 1934	M70 8/34 1538	176 2818 2674	M48 8883 1543	M90 7341 1940	K61 4509 2503	L71 4828 2391	H95 9184 1421	202 2558 2018
*	LS1 4563 2441	N58 12006	N05 5253 2346	N57 12088	H66 6723 2068	#34 6703 2081	N96 8936 1331	1483 6799 2080	N94 6687 2067	N79 12035	L53 5156 2343	N77 12132 0	170 4610 2440
I		N71 12045	M36 8123 1729	M80 7213 1970	M74 6732 2071	M59 6673 2102	M75 7336 1940	H60 6712 2090	M73 6649 2088	M84 7221 1939	M51 8039 1743	N82 12002 G	
J		L75 4582 2428	N70 12029	N93 7346 1943	M62 7365 1957	N67 12089	870 4492 2504	N76 12020	N71 7259 1954	N78 7481 1925	12026	L74 4020 2441	
ĸ			149 5084 2376	N80 12069	M34 8012 1758	144 5198 2337	LS7 5023 2394	N04 5245 2249	N53 8114 1733	N54 12024	165 5040 2363		
•				L78 4526 2432	Ne1 12196	N75 11987	N72 9238 1429	N64 12036	N78 12027	4567 2431			
м						L36 4603 2459	204 2570 2017	L76 4564 2427					

CONTENTS OF EACH CURE LOCATION

FUEL IDENTIFICATION * CURRENT U_235 GMAMS CURRENT FISSILE PU GRAMS

FIGURE 1-3

BOL BURNUP DATA

PENP UNIT 2 CYCLE 11 - START OF CYCLE BURNUP DATA - 11/14/84

7

8

9 10 11 12 13

	1	2	3	•	5	٥	. '	8	7	10		14	
						Lao	ZDI						
4							16770						
						26268	34823	26545					
					N53		865	N63					
3				15644	0	0	8895	0	0	15586			
				26246	0	0	8895	0	0	26499			
			L72	N74	N52	N07	Lá3	L58	#58		L55		
C			16915	0	12276	15849	16140	14382					
			24093	0	12276	22635	24729	23004	12131	0	24186		
		LAA	N81	802	879	N53	K39	N68	H97	#82		647	
1				8968	14937	0	6374	0	14954	14866	0	15693	
		26122	0	17392	14937	٥	28146	0	14954	14866	0	26360	
		845	877	165	164	892		891	H81	187		Néé	
ŧ		0	14564	14937	16960	17254	14764	16984	17036	15045	12094	0	
		0	14564	14937	16960	17254	14764	14984	17036	15045	12094	\$	
	1.59	-	L54	842	R67	876	187	168	163	N51	NOS	N72	Lái
	15424	0	14415	0	17283	16877	9787	14833	17242	0	15776	0	15490
	26301	o	23290	0	17283	16877	9787	14833	17242	0	22522	0	26078
	X68	161	173	K51	180	870	176	178	890	K61	11		ZB2
	14847		15983	6419	14597		5099	10014	14774	4533		8943	
	36943		24711	28243	14597	10026	39130	10014	14774	28350	24849	8943	34428
	LSI	858	-	#57		869	. 196	#83	894	N79	L53		170
	15551	0	15845	0	17075	17224	7884	16878	17143	0	14479	0	15736
	26309	. 0	22765	0	17075	17224	7884 7884	16898	17143	0	23316	0	26145
		#71	154	-	N74	#59	875	840	873	184	#51	N82	
I			12171	15232	17049	17290	14801	17139	17421	15124	12414	0	
•			12171		17069	17290	14801	17139	17421	15124	12414	0	
		L75	-	N93	847	#47	K70	#74	871	#78	N54	174	15.6
		15770					6552		15137		0		
1		26119	ő	14846	14918	ò	28435	0		14476	0	26075	
			169		154	144	L57	-	#53	N54	LAS		
v -			14994	100	12517	14359	ISALT	15811	12194	- 0	16848		
*			27802		12913	23104	24148	15811 22805	12194	0	23821		
			17945		14010	20100				1000			

L78 H41 H75 H72 H44 H78 L48 15845 0 0 8953 0 9 15489 26387 0 0 8955 0 9 26200

LS& ZB4 L76 15452 16624 15474 26307 36416 26206

CONTENTS OF EACH CORE LOCATION

FUEL IDENTIFICATION # CYCLE ASSEMBLY BURNUP TOTAL ASSEMBLY BURNUP

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PENP UNIT 2 CYCLE 11 - START OF CYCLE BURNUP DATA - 11/14/84

Section 2.0

CONTROL ROD OPERATIONAL TESTING

Cold control rod testing was conducted on November 15, 1984, just prior to initial cycle heatup.

Hot control rod testing was conducted shortly after primary system heatup on November 17, 1984.

Section 2.1 Rod Drop Times

Rod drop times to dashpot in the cold full-flow condition ranged from 1.27 seconds to 1.50 seconds with several rod drop times near each end of the range.

Rod drop times to dashpot in the hot zero flow condition ranged from 1.08 seconds to 1.18 seconds with several rod drop times near each end of the range.

Rod drop times to dashpot in the hot full-flow condition ranged from 1.23 seconds to 1.36 seconds with several rod drop times near each end of the range.

See Figures 2-1 through 2-3 for rod drop times and core parameters. Locations containing optimized fuel assemblies are marked because the narrower thimble tubes increase rod drop times slightly in the dashpot area. Locations with new control rods are also shown.

All rod drop times to dashpot were well within the Technical Specification limit of 2.2 seconds (15.3.10.E).

Section 2.2 Control Rod Mechanism Timing

Control rod mechanism timing was conducted in cold plant conditions on November 15, 1984. The visicorder traces of the lift, movable and stationary gripper coil voltages of each rod mechanism were reviewed by plant personnel. No rod misstepping occurred.

Section 2.3 Rod Position Calibration

During hot rod testing, LVDT voltages were read at 20 steps and 200 steps to determine if any voltages were abnormal.

"Zero" adjustments were made with rods at 20 steps under hot zero power full flow conditions.

"Span" adjustments were made at full power after rods were verified to be at 228 steps using WMTP 9.19. FIGURE 2-1

COLD ROD DROP TIMES (FULL FLOW)

				1					1			
+-	+	+			1.33* 1.99		1.41*					
+	+			1.35 1.91		1.48 2.09		1.39 1.94				
74			1.34						1.40			-
+[1.36 1.93		1.36 1.89				1.36 1.89		1.35 1.86		
	1.42*				1.27		1.33				1.49* 2.21	
		1.40				1.30 1.85				1.40 2.01		
	1.39* 2.04				1.32		1.27				1.45* 2.22	
-1		1.36		1.40 1.93				1.33 1.85		1.33 1.87		
-[1.38						1.34 1.91			
				1.41 1.96		1.45		1.32 1.88				
					1.50*		1.40* 2.11					
								UN	IT	2		
	١	*			ized Fi			DA	TE 11	-15-84		
			-	TIME "	NO BOTT	DM (S		TE	MP	295		°F

0*

CONTROL ROD TESTING ROD DROP TIMES FIGURE 2-2

HOT ROD DROP TIMES (NO FLOW)

				1.17*		1.17* 1.70				
			1.11 1.54		1.14		1.13			11
\geq		1.10						1.12		
-	1.10		1.10				1.09		1.11 1.54	
1.18	•			1.10		1.11 1.52				1.15
-	1.12 1.58				1.12				1.13	
1.13				1.12 1.53		1.12 1.54				1.17
_	1.12		1.09				1.08		1.12 1.54	
		1.13						1.12 1.53		
	-		1.13		1.13 <u>1.57</u>		1.11 1.56			
				1.16*		1.14*				-
								IT	2	
			Optimi	zed Fu	so* el Ass	sembly				84

0*

CONTROL ROD TESTING ROD DROP TIMES

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FIGURE 2-3

HOT ROD DROP TIMES (FULL FLOW)

+				+								
+					1.29* 1.85		1.32* 1.93					
+				1.23 1.72		1.34 <u>1.84</u>		1.32 1.82				
+	\sim		1.26 1.73						1.29 1.74			
+	-	1.28		1.25				1.26		1.27 1.75		
	1.32*				1.28 1.75		1.28 1.74				1.36* 1.98	I
		1.28				1.32 1.79				1.30 <u>1.80</u>		Ī
	1.26*				1.30 1.78		1.29 1.77				1.33* <u>1.97</u>	T
_	-	1.29		1.28				1.27 1.72		1.25		T
-	-		1.30						1.26 1.73			1
/				1.31 1.81		1.32 <u>1.81</u>		1.24				-
_					1.36		1.30* 1.90					
										2		
			-			so" nel Ass			IT	-17-84		
			=	TIME T New Co		IPOT (S DM (SE Rod		TE	MP	535	•	•
									OW	100		8

0*

CONTROL ROD TESTING ROD DROP TIMES

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Section 3.0 THERMOCOUPLE AND RTD CALIBRATION

During initial cycle heatup, thermocouple and loop RTD signals were recorded at different temperature levels under partial and full-flow conditions. See Table 3-1 for the results for full flow conditions. The RTD resistance readings were obtained at the protection racks in the control room using a digital multimeter that subtracted lead resistance. Thermocouple temperatures were read at the toggle readout panel.

Since the core was producing very little heat, the hot and cold leg RTD's were at about the same temperature. Thus both hot leg and cold leg readings were averaged into one temperature for the RTD's. The RTD resistances were converted to degrees Fahrenheit by using the vendor's calibration curves.

Due to the use of optimized fuel assemblies, the Improved Thermal Design Procedure (ITDP) was implemented. The ITDP has a requirement of 0.9% of span accuracy for the bypass manifold RTD's. To obtain the required accuracy, the existing Sostman RTD's were removed from the bypass manifolds for recalibration at PBNP. Because of poor calibration results, the Sostman RTD's were replaced with four Rosemount Model 176 and eight Model 189 RTD's.

It was found, however, that the yellow channel hot and cold leg RTD's being used were still not accurate enough as indicated in Table 3-1 and readings during initial power escalation. The spare RTD's (407A and 407B) were wired in place of 404A and 404B to obtain the required accuracy.

The T/C readout panel indicated that thermocouples at I-10, K-3, L-7, E-4, I-4, and M-6 were not functioning properly.

TABLE 3-1

RTD CALIBRATION CHECK

RTD Elements	RT	D Temperatu	res from Mea	asured Resis	stances (°F)
Loop A - Cold Leg					
R 401B	413.11	466.76	475.10	515.39	539.25
R 405B	413.48	466.67	475.42	515.56	539.05
W 402B	413.56	467.11	475.61	515.90	539.77
Loop A - Hot Leg					
R 401A	412.81	466.02	474.63	514.80	538.38
R 405A	413.98	467.09	475.75	515.84	539.30
W 402A	413.37	466.88	475.27	515.50	538.96
Loop B - Cold Leg					
B 403B	414.23	467.77	476.30	516.62	539.99
B 407B	414.02	.467.13	475.70	515.77	538.94
¥ 404B	410.55	463.53	472.10	512.27	534.83
Loop B - Hot Leg					
B 403A	413.27	466.66	475.07	515.26	538.46
B 407A	414.43	468.66	476.22	516.62	539.15
¥ 404A	414.70	468.56	476.88	517.42	540.57
RTD Average	413	467	475	516	539
T/C Average	421	479		524	545
Saturation Temp.	414	455		520	535

Section 4.0

PRESSURIZER TESTS

Section 4.1 Heater Capacity

Pressurizer heater capacity was calculated using volt and ampere readings for each group of heaters. Table 4-1 shows that heater capacity is above Technical Specification requirements of 100 KW minimum total.

TABLE 4-1

HEATER GROUP ENERGY INPUT

Group	I-Current (amps)	V-Voltage (volts)	$\frac{KW-Energy Input}{KW = \sqrt{3} \times V \times I/1000}$
A	271	480	225
В	227	480	189
С	226	480	188
D	209	480	174
E	225	480	187
		TOTAL	963

Section 4.2

Spray Valve Effectiveness

Spray value effectiveness is determined by measuring how fast each spray value decreases pressurizer pressure when fully opened with the other value closed and heaters off. For the test, spray value "A" decreased pressure at the rate of 116 psi/min. Spray value "B" decreased pressure at the rate of 113 psi/min. These are typical values and indicate that mass/flow through each value is greater than design. It can be shown that given normal heat balance characteristics of the pressurizer, 200 gpm design spray flow decreases pressure by about 70 psi/min well below the results achieved above.

Section 4.3 Heater Effectiveness

Heater effectiveness is determined by measuring how fast pressurizer pressure increases with all heaters on and spray flow only through the bypass valves. For the test, pressurizer pressure increased at an average rate of 15.6 psi/min between 1840 and 2150 psia using all heaters. This is well above design heater capacity of 14.0 psi/min. Section 5.0

REACTOR COOLANT SYSTEM

Section 5.1 RTD Manifold Flow

After the initial cycle heatup, the reactor coolant bypass flow through the RTD manifold was checked and found to be adequate for both loops. The flows were 215 gpm through Loop "A" and 190 gpm through Loop "B".

Section 5.2 Flow Transient Times

Table 5-1 gives the times to reach certain percentages of full-flow from the time a reactor coolant pump is tripped or started. The times are consistent with those obtained in previous measurements.

TABLE 5-1

REACTOR COOLANT FLOW TRANSIENT TIMES

c	ondition	Time to Reach 90% Flow (Sec.)	Time to Reach 50% Flow (Sec.)	Flow Through Active Loop (%)	Flow Through Inactive Loop (%)
A	Tripped	2.1	14.0		0
В	Tripped	2.1	14.5		0
A	Not Running			(1)	-13.8
	Started	17.5*		>108.3 ⁽¹⁾	
A	Started	18.0*		100	
B	Running			100	
A	Running			>107.9 ⁽¹⁾	
	Tripped	1.8	11.12		-17.3
A	Running			100	
B		19.6*		100	
A	Tripped	1.8	10.6		-14.3
B				107.7	

* Time to reach 100% flow.

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(1) Signal was off-scale high. Values given are for the highest scale reading.

Section 6.0

CONTROL SYSTEMS

There were no difficulties encountered during heatup or testing in the control systems of pressurizer level, pressurizer pressure, or the rod control system.

Section 7.0

TRANSIENTS

There were no significant transients during the startup or approach to full power. There were no violations of the fuel conditioning restrictions on power and rod stepping change rates.

Section 8.0 INITIAL CRITICALITY AND REACTIVITY COMPUTER CHECKS

Section 8.1 Initial Criticality

The approach to criticality was made in two phases. The first step, which began at 2020 hours on November 17, 1984, was the normal withdrawal of control rods until Bank D reached 180 steps at 2111 hours. Then the reactor coolant boron concentration was decreased by dilution until criticality was achieved. The dilution began at 2114 hours. The initial boron concentration was 1978 ppm. 10,400 gallons of water were used to reduce boron concentration by 594 ppm until criticality was achieved.

ICRR plots were maintained during each phase of the approach to criticality.

The reactor conditions at the time of criticality were determined to be as follows:

Date	November 18, 1984				
Time	0100 hours				
RCS Temperature	530°F				
RCS Pressure	1985 psig				
Rod Position	Bank D at 180 steps				
Boron Concentration	1384 ppm				

Section 8.2 Reactivity Computer Checks

- 1. Following criticality, acceptable zero power physics testing flux levels were determined. The flux level at which nuclear heat appeared was at 5×10^{-6} amps on N-35, 6×10^{-6} on N-36 and 3×10^{-6} amps on the Keithley picoammeter. Normal flux levels for physics testing are one-third of these values.
- 2. A check of the reactivity computer was made by comparing the computer's calculated reactivity for a certain doubling time versus the reactivity obtained from Figure A.1 of the WCAP. Reactor coolant system temperature was near 535°F. Table 8-1 shows the results of this check.

TABLE 8-1

REACTIVITY COMPUTER CHECKOUT

Bank D Steps		Measured	Measured	Calculated	
From	To	Doubling Time (Sec.)	Reactivity (pcm)	Reactivity (pcm)	
173	185	75.89	48	48	
172	187	57.04	59	60	
172	194	40.33	78	78	

Section 9.0 CONTROL ROD WORTH MEASUREMENT

Section 9.1 Test Description

The rod worth verification utilizing rod exchange ("rod swap") was divided into two parts. In the first part, the reactivity worth of the reference bank was obtained from reactivity computer measurements and boron endpoint data during RCS boron dilution. In the second part, the critical height of the reference bank was measured after exchange with each remaining bank.

In the rod exchange technique, the reference bank is defined as that bank which has the highest worth of all banks, control or shutdown, when inserted into the core alone. For Cycle 11 the reference bank was Control Bank A (CA) as was the case in all prior rod swap tests.

Using the analog reactivity computer, reactivity measurements were made during the insertion of Control Bank A from the fully-withdrawn to the fully-inserted position. The average current (flux level) during the measurement was approximately 5 x 10^{-7} amps. Critical boron concentration measurements (boron endpoints) were made before and after the insertion of Control Bank A (see Section 11.0). Figure 9-1 shows the results of the differential worth measurements.

Starting at a critical position with the reference bank fully inserted and Control Bank C at 212 steps, 1 new critical configuration at constant R(5 boron concentration was established with Control Bank C fully inserted and Control Bank A at 141 steps. Control Bank C was then withdrawn and Control Bank A inserted to one step to establish the initial conditions for the next exchange. This sequence was repeated until a critical position was established for the reference bank with each of the other banks individually inserted. Criticality determinations before and after each exchange were made with the reactivity computer.

The sequence of events during the rod exchange and a summary of the rod exchange data is presented in Table 9-1.

Section 9.2 Data Analysis and Test Results

The integral reactivity worth of the measured bank is inferred from the swapped portion of Control Bank A by the following equation:

$$w_X^I = w_R^M - \Delta \rho_1 - (\alpha_X) (\Delta \rho_2) + w_X^E$$
 where:

 W_X^I = The inferred worth of Bank X, pcm

- W^M_R = The measured worth of the reference bank, Control A, from fully withdrawn to fully inserted with no other bank in the core.
- $\alpha_{\rm X}$ = A design correction factor taking into account the fact that the presence of another control rod bank is affecting the worth of the reference bank.
- $\Delta \rho_2$ = The measured worth of the reference bank from the elevation at which the reactor is just critical with Bank X in the core to the reference bank fully withdrawn condition. This worth was measured with no other bank in the core.
- $\Delta \rho_1$ = The measured worth of the reference bank from the fully inserted condition to the elevation at which the reactor was just critical prior to the worth measurement of Bank X. In this test $\Delta \rho_1$ is zero.
- W_X^E = The worth of Bank X from the initial position (before the start of the exchange) to 228 steps. This worth is measured by the normal endpoint worth method.

Final values for the integral worth of control and shutdown banks inferred from the measurement data are tabulated in Table 9-2. Values for α_{ν} were obtained from the design predictions are also listed in Table 9-2.

Section 9.3 Evaluation of Test Results

A comparison of the measured/inferred bank worths with design predictions is presented in Table 9-2.

In evaluating the test results, the standard review and acceptance criteria were used.

Review Criteria

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- a. The measured worth of the reference bank agrees with design predictions within ±10%.
- b. The inferred individual worth of each remaining bank agrees with design predictions within ±15% or ±100 pcm whichever is greater.
- c. The sum of the measured and inferred worths of all control and shutdown banks is less than 1.1 times the predicted sum.

Acceptance Criteria

a. The sum of the measured/inferred worths of all control and shutdown banks is greater than 0.9 times the predicted sum.

As shown on Table 9-2, all review and acceptance criteria were met.

CONTROL BANK A WORTH

PENP UNIT 2 CYCLE 11 BOL HZP

All Other Rods Fully Withdrawn

Steps Withdrawn

Differential Worth (PCM/STEP)

1

0 - Measured Data Solid Line - Design

50100 5 2

TABLE 9-1

. 1

Bank Measured	Time	RCS Tavg (°F)	CA Position (Steps)	Measured Bank Position (Steps)
cc	1721	530	1	212
CC	1732	530	141	1
CC	1744	531	1	214
SB	1747	531	1	217
SB	1757	531	131	1 .
SB	1808	531	1	217
SA	1819	531	1	213
SA	1827	531	127	1
SA	1839	531	1	214
CD	1849	531	1	213
CD .	1856	531	83	1
CD	1905	531	1	220
CB	1906	531	1	218
CB	1912	530	101	1
CB	1920	530	1	220

CRITICAL ROD CONFIGURATION DATA

Boron concentration was 1190 ppm.

TABLE 9-2

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CC	530	0.956	37	1125	1149	- 2.1	
SB	602	1.009	23	1011	996	+ 1.5	
SA	631	0.953	23	1017	993	+ 2.4	
CD	1014	0.991	24	. 607	580	+ 5.9	
CB	837	1.077	27	714	690	+ 3.7	
CA				1595	1650	- 3.3	
		T	TAL	6076	6058	+ 0.3	

Section 10.0 TEMPERATURE COEFFICIENT MEASUREMENTS

Isothermal temperature coefficient measurements were taken during zero power physics testing. The measurement test conditions and results are given in Table 10-1. The measured values are the average of the recorded reactor coolant system heatups and cooldowns. Reactivity from the reactivity computer and reactor coolant system temperature were recorded on an X-Y plotter and two-pen recorder.

The measured temperature coefficients are within the review criteria of ±3 pcm/°F.

TABLE 10-1

ISOTHERMAL TEMPERATURE COEFFICIENTS

Control Bank Configuration	Boron Avg. Conc. Temp ppm °F		Measured pcm/°F	Design* pcm/°F	Difference pcm/°F (M-D)	
ARO	1351	534	-3.1	-3.1	0.0	
A in	1189	530	-7.3	-6.3	-1.0	

*WCAP Figure 5.1 and Figure 5.8

Section 11.0 BORON WORTH AND ENDPOINT MEASUREMENTS

Figure 11-1 shows RCS boron concentration during zero power physics testing. Table 11-1 shows results of the endpoint measurements. Design values are for 530°F testing temperature. The measured boron worth was obtained by dividing bank worth (pcm) into change in boron concentration between the endpoints.

Review criterion was not met $(\pm 0.5 \text{ pcm/ppm})$. This is a typical problem with boron endpoint measurements where measured boron endpoints are not close to design.

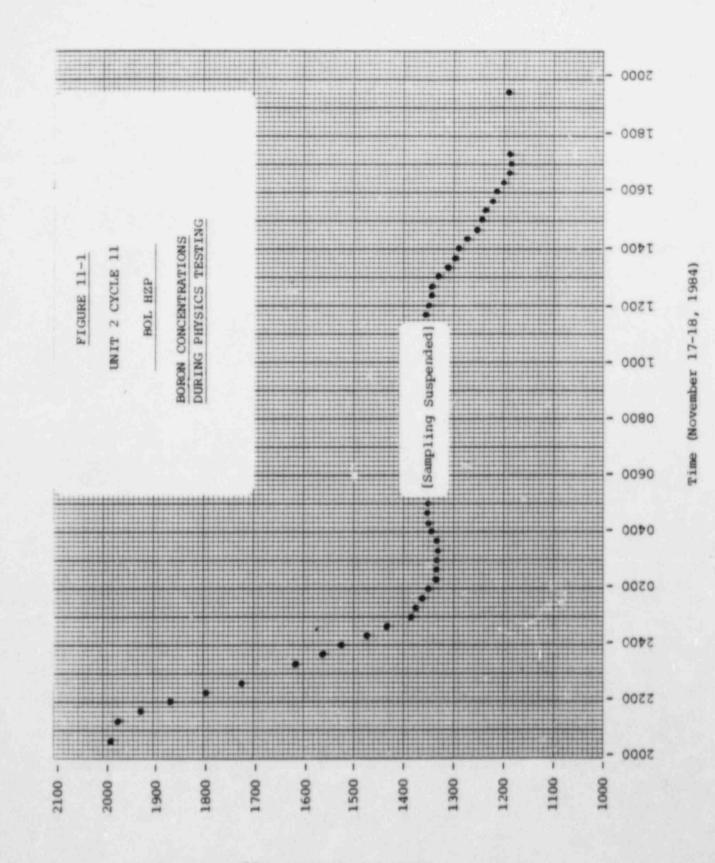
TABLE 11-1

BORON WORTH AND ENDPOINTS

		oint	Bank Worth		Boron Worth	
Bank Configuration	Design ⁽¹⁾ (ppm)	Measured (ppm)	Design (pcm)	Measured (pcm)	Design ⁽²⁾ (pcm/ppm)	Measured (pcm/ppm)
ARO	1373	1355			-9.19	
CA in	1189	1192	1650	1595	-9.15	-9.8

(1) Figure 5.1 Table A.2

(2) Figure 2 - Supplement to WCAP, Letter 84WE-G-080 Table A.2



BOYON CONCENTIALION (PPM)

29

Section 12.0 POWER DISTRIBUTION

Table 12-1 illustrates the lowering of maximum hot channel factors during initial power increase to full load. More flux maps were required because allowed power levels based on maximum hot channel factors were less than 100% for the HZP flux map. Allowed power levels were calculated using the relationships for F Δ H and FQ versus power level in Technical Specification 15.3.10.B.1.a. The relationships have been changed due to the use of optimized fuel assemblies in Cycle 11. Zero power flux map results typically do not show that full power operation is permitted due to hot channel factor limitations.

Measured power sharing factors (F Δ H) for each fuel assembly were compared to predicted values. Differences of more than 5% were listed in Figure 12-1 for the ARO HZP flux map and for a full power map (No. 9) taken after a month of operation.

Figures 12-2 and 12-3 show the actual power sharing factors at each location for the same flux maps.

Measured axial power distribution compared to design is shown in . Figures 12-4 and 12-5 for the same flux maps.

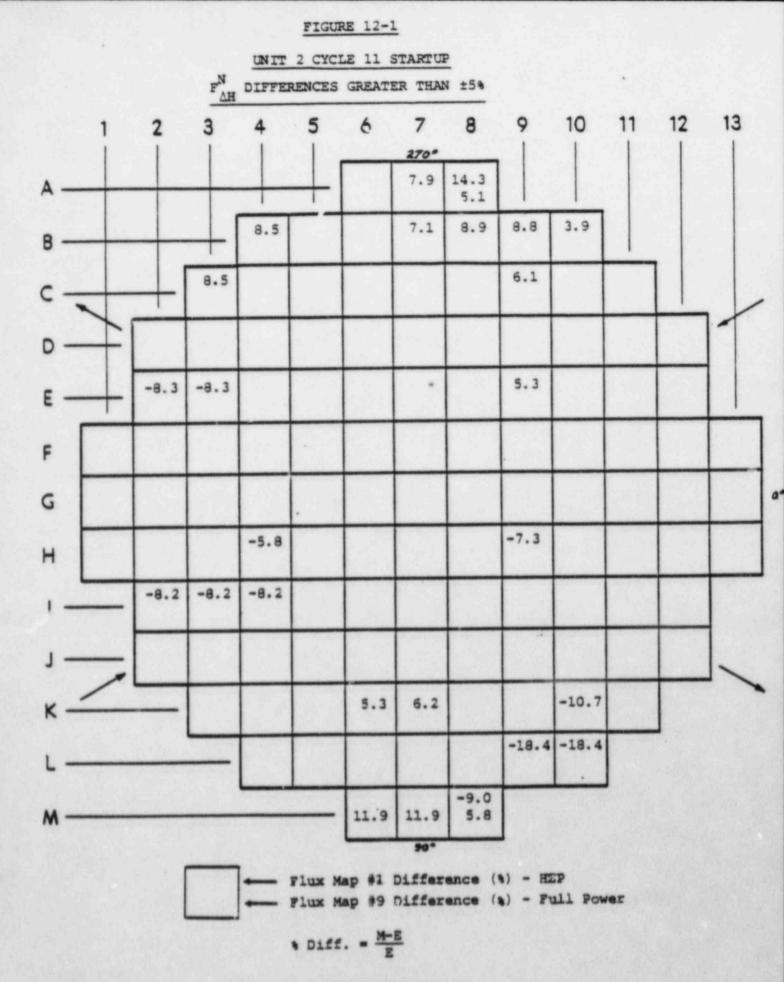
TABLE 12-1

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INITIAL POWER ESCALATION FLUX MAP RESULTS

Flux Map Number	Date	Power	Thimbles Missing	Allowed Power (%)		
				FAHN	FQN	
1	11-18-84	0	5	100	76	
2	11-20-84	20	1	112	104	
3	11-26-84	50	1	117	113	
* 4	11-27-84	50	2	114	109	
* 5	11-27-84	50	1	116	107	
6	12-03-84	100	1	116	118	
* 7	12-04-84	100	1	116	115	
* 8	12-04-84	100	1	116	115	
9	12-19-84	100	1	117	119	

* QAO flux maps taken when delta flux or control rods were not near their normal operating positions.



POINT BEACH NUCLEAR PLANT

FIGURE 12-2

POWER DISTRIBUTION, HZP, ARO

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PBFN 211-14 HZP AR0 . 11/18/84

	P.	2	3	٠	5	ó	7	8	,	10	11	12	13
							0.284						
1						1.2		14.3					
					0.936					0.436			
3				0.443	0.947		1.016			0.453			
			0.470	1.083	1.153	0.996	0.893	0.798	1.216	1.183	0.519		
C			0.510				0.916		10 000000000000000000000000000000000000	1.229	and the second second		
			8.5	3.6	-0.2	0.1	2.6	4.4	á.1	3.9	1.5		
		0.403	1.119	1.113	1.181	1.136	0.958	1.154	1.276	1.313	1.213	0.456	
3		0.403	1.123		1.185								
		-0.1	0.3	3.4	0.3	0.5	2.9	4.3	5.0	3.5	2.0	1.5	
		0.905	1.113	1-145	1.199	1.219	1.744	1.251	1.294	1.311	1.284	1.074	
ε			1.021		1.199								
		-8.3	-8.3		-0.0			4.5	5.3		2.4		
		1.074			1.199						1.086		
F			0.919	1.083	1.204	1.234	1.261	1.244	1.282	1.202		1.252	0.334
	-3.7	-3.7	-3.4	-2.2	9-4	1.7	4.8	-0.8	-9.3	-1.4	3.1	3.9	4.1
	0.315	0.756	0.894	0.754	1.234	1.201	0.883	1.235	1.293	1.016	0.963	1.029	0.316
6	0.317	0.924	0.862.	0.915	1.231							1.049	
	0.7	-3.3	-3.4	-4.1	-0.2	1.5	2.4	-2.4	-3.8	-4.7	-1.0	3.9	4.1
	0.321	1.144	1.035		1.252				1.290		1.058	1.184	0.334
н	10.00	1.101		1. (m) 1. (m) 1. (m)	1.254						1.033		
	0.7	-3.7	-3.9	-5.8	9.1	2.0	-0.1	-4.9	-7.3	-4.5	-2.1	1.4	4.1
		0.994			1.281							1.023	
1		0.913	1.136	1.175		1.245	1.309	1.325			1.217		
		-9.2	-0.2	-8.2	0.4	0.4	1.1	1.2	-1.7	-4.4	-4.4	-3.5	
		0.442	1.184	1.297				1.229			1.224		
1			1.236		1.332			and the second se		1.299		and the second sec	
		4.2	4.2	4.2	4.2	2.8	4.3	4.5	1.9	-3.4	-3.4	-3.5	
				1.181									
*				1.210									
			2.4	2.5	2.4	5.3	0-2	-1.7	-5.1	-10.7	-1.1		
				0.439		1.150	1.002		1.031	0.458			
5				0.449	1.013	1.179	1.028	1.131	0.841	1.00.00.000			
				2.3	2.3	2.4	2.6	-4.7	-18.4	-18.4			
						0.325	0.300	0.333					
						0.324	0.300	0.393					
						-0.1	-0.1	-9.0					

PSFH 211-1A HZP ARO 11/18/84

0.000 PREDICTED FDHM 0.000 MEASURED FDHM 0.0 I DIFF. (M-P)/P

0.0 I FZ = 1.485 FDHM = 1.520 AT F12DE F0 = 2.545 AT F12DE

FIGURE 12-3

POWER DISTRIBUTION AT POWER

PBFH 211-9 1500.9 MUT ROUTINE 12/19/84

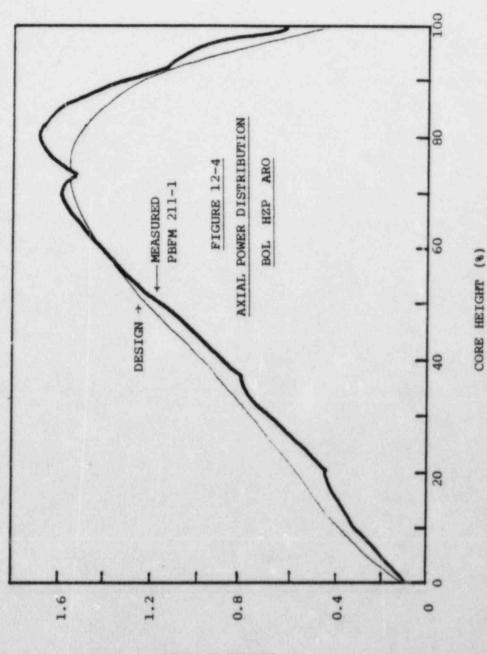
	1	2	2	•	5	4	7	8	9	10	11	12	13
						0.336	0.334	0.355					
A						0.354	0.343	0.373					
						-0.3	2.4	5.1					
					0.984								
8					0.980								
				2.1	-0.3	-0.3	2.4	3.4	3.6	2.0			
					1.183					1.164			
C			0.528		1.182					1.184	100 C		
			2.1	1.4	-0.0	9.4	1.1	2.3	2.8	2.0	0.9		
			1.160		1.201					1.258			
0			1.176		1.198						1.180		
		1.3	1.4	1.2	-0.3	0.2	1.0	2.3	1.7	1.0	0.1	0.9	
		0.954	1.147	1.188	1.209	1.219	1.233	1.227	1.251	1.258	1.235	1.021	
ε		0.962	1.154		1.207								
		0.4	0.4	0.2	-9.2	0.1	0.1	0.5	-0.5	-0.7	-1.4	-2.1	
8	0.350	1.125	1.002	1.142	1.203	1.207	1.189	1.221	1.241	1.197	1.072	1.184	0.369
F		1.140	1.010	1.142	1.203	1.207	1.189	1.216	1.235	1.182	1.053		0.364
	1.4	1.4	0.8	-0.1	-0.0	0.0	-0.1	-4.4	-0.5	1.3	-1.3	-1.7	-1.3
6	0.340	1.006	0.957	0.988	1.228	1.189	0.897	1.201	1.247	1.011	0.963	1.035	0.332
8			0.935						1.232	0.992	0.941	1.017	0.347
	0.7	-0.1	-9.2	-4.7	0.2	0.5	-0.3	-0.9	-1.2	-1.8	-2.3	-1.7	-1.3
		1.148		1.181	1.230	1.221	1.208	1.234			1.045	1.149	0.345
H		1.145	1.054										
	0.4	-0.2	-9.2	-1.1	0.3	0.5	-4.3	-9.8	-1.4	-1.6	-1.8	-1.2	-0.6
1			1.219					and the second se	1.259			1.004	
I		0.993	1.197	1.226	1.248		1.242		1.240		1.198		
		-1.4	-1.8	-1.8	9.2	0.2	-4.7	-0.8	-1.5	-1.4	-1.4	-1.5	
		0.448			1.241				1.259		1.175		
3		0.460			1.235					1.251			
		-1.7	-1.1	-1.1	-0.5	-1.0	-0.9	-0.9	-1.0	-1.0	-1.4	-1.4	
					1.204			1.043		1.176			
ĸ					1.192								
			-1.2	-1.2	-1.0	-2.2	-1.4	-1+1	-0.9	-1.0	-1.3		
					0.994								
6					0.784		1.069	Contraction (Sec.)	1.004				
				-1.0	-1.0	4.9	4.9	3.0	-9.6	-0.0			
						4.159	0.170	A 747					

0.358 0.339 0.343 0.401 0.379 0.384 11.9 11.9 5.8

PBFN 211-9 1500.9 MUT ROUTINE 12/19/84

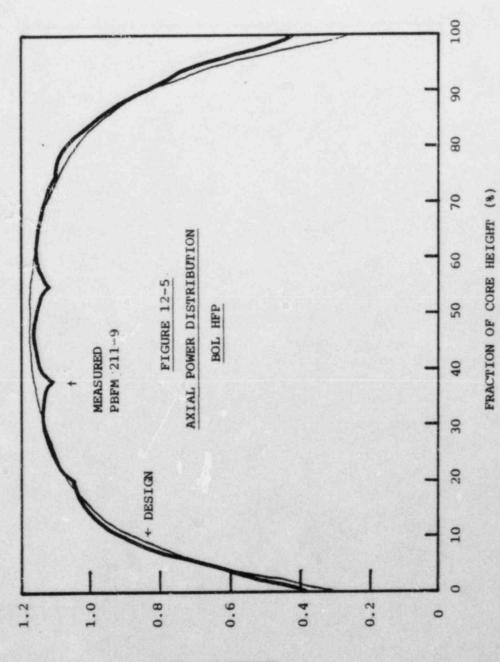
CONTROL ROD POSITIONS (STEPS) 0.000 PREDICTED FDHM SHBUN SHBUN CNTRL CNTRL CNTRL CNTRL 0.000 MEASURED FDHM A B A B C B 0.0 I DIFF. (M-P)/P 223. 223. 228. 228. 213.

POWER 1501. NUT 98.8 2 FZ = 1.161 FDHN = 1.443 AT L SJD FG = 1.698 AT L SJD



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RELATIVE POWER



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RELATIVE POWER

Section 13.0

XENON REACTIVITY

Xenon reactivity behavior data for Unit 2 Cycle 11 was supplied by Westinghouse as part of the WATCH data package. Point Beach code XENALG will be run with a TDF1 of 0.95 and a TDF2 of 1.2 to remain consistent with the Xenon Tables. Tables are supplied for BOL, MOL and EOL conditions.

Section 14.0

SHUTDOWN MARGIN CONSIDERATIONS

Rod swap results were within acceptance criteria and were accepted as valid proof of rod worth for shutdown margin determination. See Section 9.0 for rod swap details. Thus WCAP Table 6.3 was accepted as a valid shutdown margin determination. Table 14-1 calculates the excess worth available to Unit 2 Cycle 11.

TABLE 14-1

EXCESS SHUTDOWN WORTH AVAILABLE FOR A FULL POWER TRIP

Chutdam Margin	BOL (pcm)	EOL (pcm)
Shutdown Margin From WCAP Table 6.3	-4230	-3500
- Required Shutdown	-1000	-2770
= Excess Worth	-3230	- 730

Section 15.0

EXCORE DETECTOR BEHAVIOR

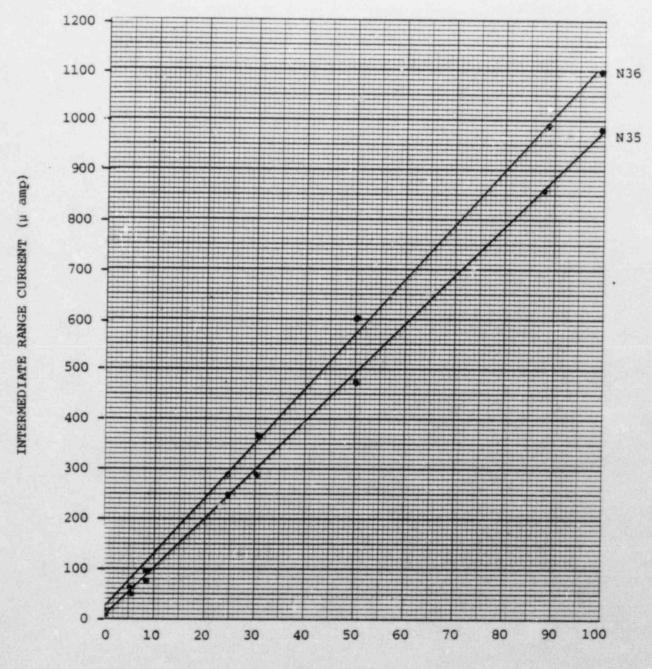
Section 15.1 Excore Detector Current Versus Power Level

The upper and lower excore detector currents for each power range channel were recorded and calorimetrics were performed at various power levels. The upper and lower detector currents were summed for each channel and then normalized to obtain predicted currents for 100% power. These 100% currents are listed in Table 15-1.

Intermediate range detector currents versus power level are shown in Figure 15-1. The intermediate range detector trip signals activated at about 2.9 x 10^{-4} amps and 3.2 x 10^{-4} amps for N35 and N36 respectively. From Figure 15-1, the trip signals occurred between 28% and 30% power as expected.

FIGURE 15-1

INTERMEDIATE RANGE DETECTOR RESPONSE TO POWER LEVEL



POWER LEVEL (%)

	100%	CURRENTS	(µ AMPS)	
	<u>41</u>	42	<u>43</u>	44
Cycle 11	576	626	379	573
Cycle 10	625	641	403	574
Cycle 9	615	651	415	610

TABLE 15-1

Section 15.2 Excore Axial Offset Response

Excore axial offset responds to actual incore axial offset (calculated from flux map data) in a linear fashion but not with a one-to-one correspondence. Excore axial offset "sees" only about 70% of the actual axial offset. Table 15-2 shows the historic response of the detectors.

Section 15.3 Channel Calibration

The currents measured during flux maps at three different axial offsets were first corrected for quadrant tilt by dividing each channel current by the quadrant tilt factor calculated by PBCORE for the associated quadrant. Then because each flux map was taken at a slightly different power level, the currents from two maps were ratioed up or down by the percent the power level had changed from the reference map.

Straight line fits of the "corrected" currents for each channel versus incore axial offset as determined by the flux maps were obtained. The intersection of the line with zero axial offset was the calibration current at the power level of the reference map.

Power range quadrant tilt alarms were meant for rapidly developing tilts. Natural core tilts were washed out to prevent a bias in alarming of rapidly occurring tilts. This was accomplished by multiplying the calibration currents for each channel by the quadrant tilt factors calculated by PBCORE, to obtain "tilt free" calibration currents. Thus, after the "tilt free" calibration currents were entered, the computer and the Hagan recorders indicated the same power (voltage) on all upper half quadrants and the same power on all lower half quadrants.

In actual practice, the "tilt free" calibration currents are ratioed to a power level slightly above normal operating full power to make it possible for the currents to be set in at power. Table 15-3 lists the actual "tilt free" calibration currents at 100% power at BOL.

TABLE 15-2

EXCORE AXIAL OFFSET RESPONSE HISTORY

		Slope (Incore	vs. Exc	core)
		<u>41</u>	<u>42</u>	<u>43</u>	44
Cycle 1	11	1.45	1.37	1.17	1.38
Cycle 1		1.56	1.55	1.27	1.50
Cycle 9	9	1.49	1.58	1.27	1.66

TABLE 15-3									
	BOL	CALIBRATION	CURRENTS	(100%)					
	41	<u>42</u>	<u>43</u>	<u>44</u>					
	304	321	201	290					
	265	291	168	271					

T

В

Section 16.0 OVERPOWER, OVERTEMPERATURE AND DELTA FLUX SETPOINTS CALCULATION

Section 16.1 Overpower and Overtemperature AT Setpoints Calculation

Discussion of the setpoints and equations has been sufficiently covered in previous reports.

The equations are:

Overpower $\Delta T \left(\frac{1}{1+\tau_3 S} \right)$

$$\leq \Delta T_{o} \left[K_{4} - K_{5} \left(\frac{\tau_{5}S}{\tau_{5}S + 1} \right) \left(\frac{1}{1 + \tau_{4}S} \right) T - K_{6} \left[T \left(\frac{1}{1 + \tau_{4}S} \right) - T' \right] - f(\Delta I) \right]$$

Overtemperature $\Delta T(\frac{1}{1+\tau_3 S})$

$$\leq \Delta T_{o} (K_{1} - K_{2}(T(\frac{1}{1+\tau_{4}S}) - T^{1})(\frac{1+\tau_{1}S}{1+\tau_{2}S}) + K_{3} (P-P^{1}) - f(\Delta I))$$

See Tables 16-1 and 16-2 for the constants associated with this cycle of operation.

Section 16.2 Delta Flux Setpoints Calculation

The overpower and overtemperature ΔT setpoints are reduced when the excore detectors sense a power mismatch between the top and bottom of the core. The dead band is +5% and -17% before the setpoints are reduced. For each percent (more than 5%) the top detector output exceeds the bottom detector, the setpoints are reduced an equivalent of 2% of the rated power. For each percent (more than -17%) the bottom detector exceeds the top detector, the setpoints are reduced an equivalent of 2% of rated power.

TABLE 16-1

OVERTEMPERATURE AT CONSTANTS

 ΔT_{a} = Indicated ΔT at rated power, °F

T = Average temperature, °F

 $T^1 = 574.2^{\circ}F$

P = Pressurizer pressure, psig

 $P^1 = 2235 psig$

 $K_1 ≤ 1.117$ for operation at 2250 psia primary system pressure ≤1.30 for operation at 2000 psia primary system pressure $K_2 = 0.0150$

 $K_3 = 0.000791$

 $\tau 1 = 25$ seconds

 $\tau 2 = 3$ seconds

 $\tau 3 = 2$ seconds for Rosemount or equivalent RTD

= 0 seconds for Sostman or equivalent RTD

t4 = 2 seconds for Rosemount or equivalent RTD

= 0 seconds for Sostman or equivalent RTD

TABLE 16-2

OVERPOWER AT CONSTANTS

ΔT _o	=	Indicated ΔT at rated power, °F					
т	=	verage temperature, °F					
T'	=	574.2°F					
K4	<#	1.089 of rated power					
K ₅	=	0.0262 for increasing T					
	=	0.0 for decreasing T					
K ₆	=	0.00123 for $T \ge T$					
	=	0.0 for T < T'					
τ5	=	10 seconds					
		$f(\Delta I)$ as defined in Section 16.2					
τ3	=	2 seconds for Rosemount or equivalent RTD					
	=	0 seconds for Sostman or equivalent RTD					

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τ4 = 2 seconds for Rosemount or equivalent RTD

= 0 seconds for Sostman or equivalent RTD

Section 17.0

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FUEL PERFORMANCE

Reactor coolant activity is summarized in Table 17-1 and indicates good fuel integrity.

Because of low Cycle 10 activity, no fuel assembly failures were expected. Thus, there were no fuel assembly inspections scheduled other than the demonstration optimized fuel assemblies. Those assemblies were in good condition.

TABLE 17-1

TYPICAL ISOTOPIC COMPOSITION OF PRIMARY COOLANT ACTIVITY

Isotope	Half Life	End of Cycle 10 µC/cc x 10 ⁻¹	Start of Cycle 11 _µC/cc x 10 ⁻¹
I-131	8.05 days	0.1	0.0
I-132	2.3 hours	1.5	0.7
I-133	21 hours	1.0	0.5
I-134	53 minutes	2.5	1.2
I-135	6.7 hours	2.0	<u>1.0</u>
	TOTAL	7.1	3.5
	ivity (µCi/cc) te decay	0.6	0.3

Section 18.0 CONCLUSION

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The use of optimized fuel assemblies produced no unusual physics testing results. The use of optimized fuel assemblies had no significant effects on other phases of startup testing.