

WCAP-10167

WESTINGHOUSE CLASS 3

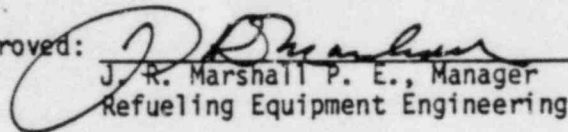
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EVALUATION OF THE ACCEPTABILITY OF THE REACTOR  
VESSEL HEAD LIFT RIG, REACTOR VESSEL INTERNALS  
LIFT RIG, LOAD CELL, AND LOAD CELL LINKAGE  
TO THE REQUIREMENTS OF NUREG 0612  
for  
PUBLIC SERVICE ELECTRIC AND GAS COMPANY  
SALEM GENERATING STATION  
UNITS 1 AND 2

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## ABSTRACT

An evaluation of the Salem reactor vessel head and internal lift lugs, load cell, and load cell linkage was performed to determine the acceptability of these devices to meet the requirements of NUREG 0612. The evaluation consists of: (1) a comparison report of the ANSI N14.6 requirements and the requirements used in the design and manufacture of these devices; (2) a stress report in accordance with the design criteria of ANSI N14.6; and (3) a list of recommendations to enable these devices to demonstrate compliance with the intent of NUREG 0612 and ANSI N14.6.



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## ATTACHMENTS

- A. Comparison of ANSI N14.6-1978 Requirements for Special Lifting Devices and the Requirements for the Reactor Vessel Head Lift Rig, Reactor Vessel Internals Lift Rig, Load Cell, and Load Cell Linkage for Public Service Electric and Gas Company, Salem Generating Station, Units 1 and 2.
  
- B. Stress Report - Reactor Vessel Head Lift Rig, Reactor Vessel Internals Lift Rig, Load Cell, and Load Cell Linkage for Public Service Electric and Gas Company, Salem Generating Station, Units 1 and 2.

## REFERENCES

1. George, H., Control of Heavy Loads at Nuclear Power Plants Resolution of Generic Technical Activity A-36, NUREG 0612, July, 1980.
2. ANSI N14.6-1978 Special Lifting Devices for Shipping Containers Weighing 10,000 Pounds or More for Nuclear Material
3. ANSI B 30.9-1971. Slings, American National Standards Institute, New York, 1971.
4. Westinghouse Drawing 685J059 - 3 Loop Head Lifting Rig General Assembly
5. Westinghouse Drawing 113E480 - 4 Loop Plants Lifting Rig Internals - General Assembly

SECTION 1  
INTRODUCTION

The Nuclear Regulatory Commission (NRC) issued NUREG 0612 "Control of Heavy Load at Nuclear Power Plants"<sup>[1]</sup> in 1980 to address the control of heavy loads to prevent and mitigate the consequences of postulated accidental load drops. NUREG 0612 imposes various training, design, inspection and procedural requirements for assuring safe and reliable operation for the handling of heavy loads. In the containment building, NUREG 0612 Section 5.1.1(4) requires special lifting devices to meet the requirements of ANSI N14.6-1978-"American National Standard for Special Lifting Devices for Shipping Containers Weighing 10,000 Pounds or More for Nuclear Materials".<sup>[2]</sup> In general, ANSI N14.6 contains detailed requirements for the design, fabrication, testing, maintenance, and quality assurance of special lifting devices. In addition, ANSI N14.6 requires that when wire rope or chain is used in the design of a special lifting device, the wire rope or chain shall be in conformance with ANSI B30.9-1971 "American National Standard Safety Standard for Slings".<sup>[3]</sup> The Salem Generating Station lifting devices which can be categorized as special lifting devices and which are contained in the scope of this report are:

1. Reactor vessel head lift rig
2. Reactor vessel internals lift rig
3. Load cell
4. Load cell linkage

This report contains the evaluation performed on these lifting devices to determine the acceptability of these devices to meet the above requirements.



## 1.1 BACKGROUND

The reactor vessel head lift rig, the reactor vessel internals lift rig, load cell, and load cell linkage were originally designed and built for the Salem Generating Station circa 1967-71. Subsequently, the Unit 2 internals lift rig was transferred to another plant site in 1975-76 and replaced in 1977 with an almost identical design. The actual design criteria is unknown for the lifting devices. It appears that Westinghouse used the design criteria that the resulting stress in the load carrying members, when subjected to the total combined lifting weight, should not exceed one fifth (1/5) of the ultimate strength of the material. These items were not classified as nuclear safety components and requirements for formal documentation of design requirements and stress reports were not applicable. Thus, stress reports and design specifications were not formally documented. Westinghouse defined the design, fabrication and quality assurance requirements on detailed manufacturing drawings and purchase order documents. Westinghouse also issued field assembly instructions for the reactor vessel head and internals lift rigs which included an initial load test followed by non-destructive surface examination of critical load bearing areas. Additionally, the new Unit 2 internals lift rig was load tested at the manufacturers shop at 125 percent of the maximum weight. Subsequent modifications (presently in progress) to the internals lifting rig include roto-lock studs and inserts designed, manufactured, and tested to the requirements of ANSI N14.6.



SECTION 2  
COMPONENT DESCRIPTION

2.1 REACTOR VESSEL HEAD LIFT RIG

The reactor vessel head lift rig<sup>[1]</sup> is a three-legged carbon steel structure, approximately 43 feet high and 14 feet in diameter, weighing approximately 28,000 pounds. It is used to handle the assembled reactor vessel head.

The three vertical legs and Control Rod Drive Mechanism (CRDM) platform assembly are permanently attached to the reactor vessel head lifting lugs. The tripod sling assembly is attached to the three vertical legs and is used when installing and removing the reactor vessel head. During plant operations, the sling assembly is removed and the three vertical legs and platform assembly remain attached to the reactor vessel head.

2.2 REACTOR VESSEL INTERNALS LIFT RIG

The reactor vessel internals lift rig<sup>[2]</sup> is a three-legged carbon and stainless steel structure, approximately 25 feet high and 14 feet in diameter weighing approximately 14,500 pounds. It is used to handle the upper and lower reactor vessel internals packages. It is attached to the main crane hook for all lifting, lowering and traversing operations. A load cell linkage is connected between the main crane hook and the rig to monitor loads during all operations. When not in use, the rig is stored on the upper internals storage stand.

The original rig design attaches to the internals packages by means of three engaging screws which are screwed into tapped holes in the internals flanges. These screws are manually operated from the manipulator crane walkway using a handling tool which is essentially a long wrench. The screws are normally spring retracted upward and are depressed to engage the tapped holes by the weight of the handling tool. Modifications to these rigs, presently in progress, include installation of an

operating platform, integral tools and a change to the type of engagement with internals to a rotolock stud and insert type. These rotolock studs are manually operated from the new internals lift rig platform using a handling tool which is an integral part of the rig. The studs are normally spring retracted upward and are depressed to engage the inserts. Rotating the mechanism locks it in both positions. Since these rotolock studs are in the process of being incorporated, both types of load carrying items are included in this report.

### 2.3 LOAD CELL AND LOAD CELL LINKAGE

The load cell is used to monitor the load during lifting and lowering the reactor vessel head or internals to ensure no excessive loadings are occurring. It is installed between the load cell linkage and the lifting device. The load cell is a strain gage (tension) type, rated at 400,000 pounds and built by W. C. Dillon and Co. The load cell linkage is an assembly of pins, plates and bolts which connect the polar crane main hook to the load cell.

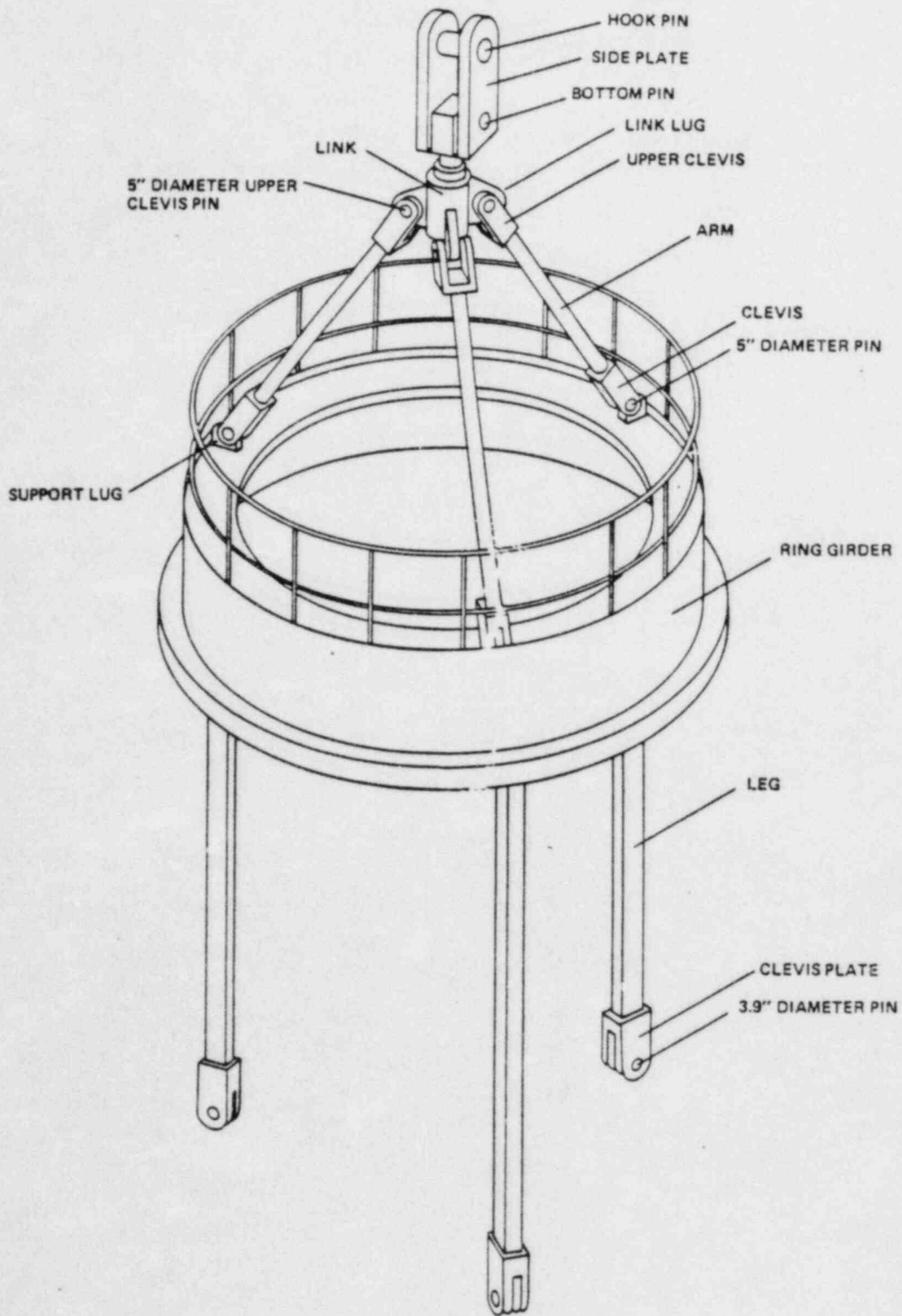


Figure 2-1. Reactor Vessel Head Lift Rig

WHERE PSE AND PNJ RIGS DIFFER, THE PSE ITEMS ARE DISTINGUISHED BY AN "A" SUFFIX, THE PNJ ITEMS BY A "B" SUFFIX

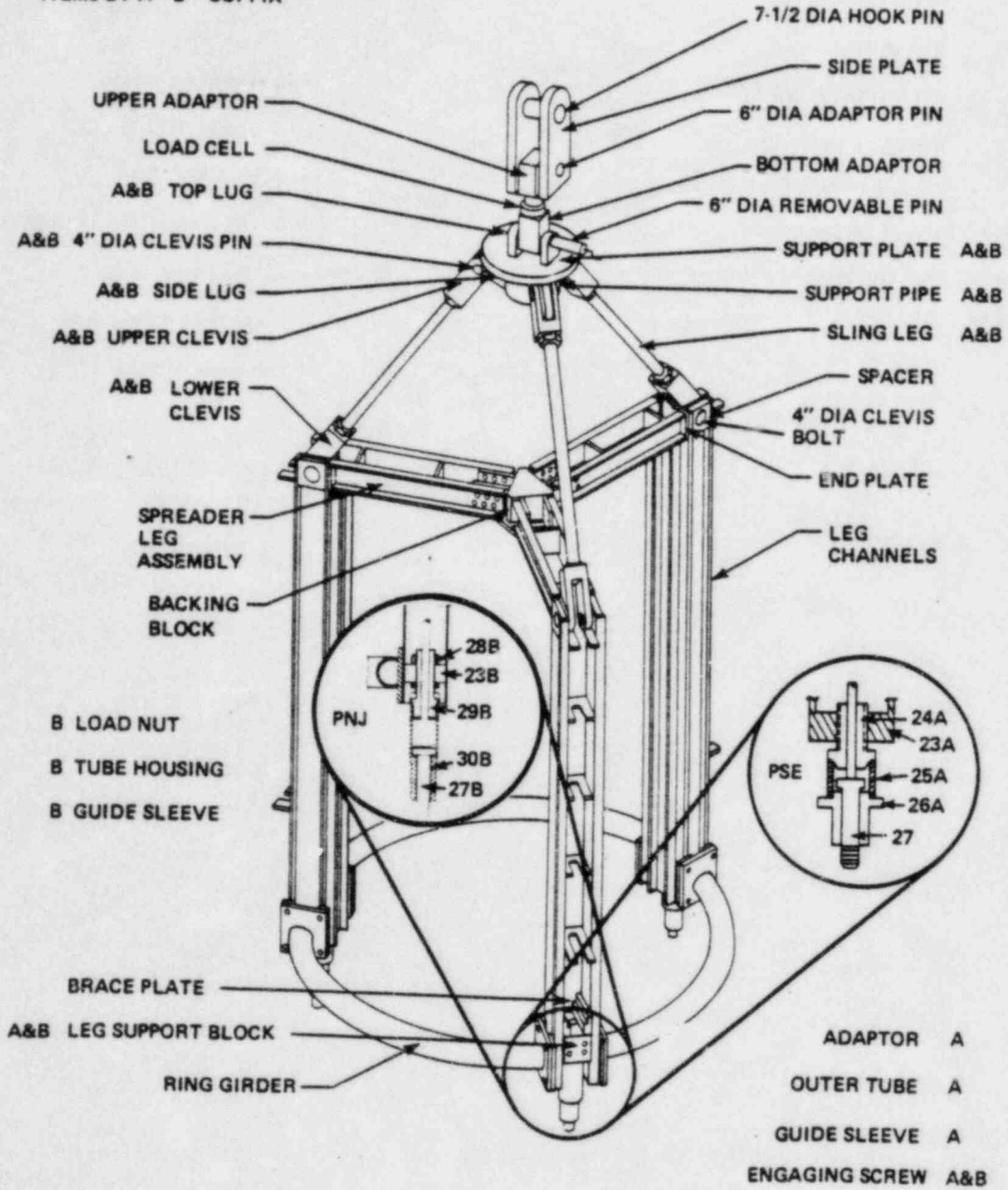


Figure 2-2. Reactor Vessel Internals Lift Rig

SECTION 3  
SCOPE OF EVALUATION

The evaluation of these lifting devices consists mainly of three parts:

1. A detailed review of the ANSI N14.6 requirements
2. Preparation of a stress report
3. Recommendations to demonstrate compliance with NUREG 0612, Section 5.1.1(4).

Discussion of these items follows.

### 3.1 REVIEW OF ANSI N14.6-1978

A detailed comparison was made of the information contained in ANSI N14.6 with the information that was used to design, manufacture, inspect and test these special lifting devices. The detailed comparison is provided in three parts:

1. Overall item by item comparison of requirements
2. Preparation of a critical item list per ANSI N14.6 Section 3.1.2, and
3. Preparation of a list of nonconforming items.

This detailed analysis is contained in Attachment A to this report.

### 3.2 PREPARATION OF A STRESS REPORT

Section 3.1.3 of ANSI N14.6 and NUREG 0612 Section 5.1.1(4) require a stress report to be prepared. Special loads and allowable stress criteria are specified for this analysis. The stress report is Attachment B to this report.



### 3.3 RECOMMENDED ACTIONS

An obvious result from the previous evaluations is a list of items that can be performed to demonstrate to the NRC that these special lifting devices are in compliance with the guidelines of ANSI N14.6 and NUREG 0612 Section 5.1.1(4). These recommendations are identified in Section 6.



SECTION 4  
DISCUSSION OF EVALUATIONS

4.1 STUDY OF ANSI N14.6-1978

A review of ANSI N14.6 identifies certain analyses to be performed and certain identifications that are required to be made to demonstrate compliance with this document. These are a preparation of a stress report in accordance with Section 3.2 and a preparation of a critical items list in accordance with Section 3.1.2. The stress report is Attachment B to this report. The critical items list has been prepared per Section 3.1.2 and is contained in Appendix A to Attachment A. This list identifies the critical load path parts and welds, the materials of these items, and the applied non-destructive volumetric and surface inspections that were performed. (Details of these non-destructive processes and acceptance standards are available at Westinghouse should they be needed.)

A detailed item by item comparison of all the requirements of ANSI N14.6 and those used for the design, manufacture and inspection of these lifting devices is contained as Table 2-1 of Attachment A. The comparison shows that these devices meet the intent of the ANSI document for design, fabrication and quality control. However, they do not meet the requirements of ANSI N14.6 for periodic maintenance, proof and functional testing. Thus, a tabulation of those ANSI N14.6 requirements that are incompatible with these lifting devices was prepared and is Appendix B to Attachment A. Included in Appendix B to Attachment A are recommended actions that may be used to demonstrate acceptability to the NRC.

4.2 STRESS REPORT

As part of the invoking of the ANSI N14.6 document, the NRC requested utilities to demonstrate their compliance with the stress criteria with some qualifying conditions. Attachment B is the stress report for these devices performed in accordance with the criteria of ANSI N14.6. A

discussion is included which responds to the NRC qualifying conditions of NUREG 0612. All of the tensile and shear stresses meet the design criteria of Section 3.2.1.1 of ANSI N14.6, requiring application of stress design factors of three and five with the accompanying allowable stress limits of yield and ultimate strength, respectively.

#### 4.3 RECOMMENDATIONS

The recommendations identified in Section 6 require a review of plant maintenance and operating instructions to ensure that they contain information relative to the identification, maintenance and periodic testing required by ANSI N14.6. The extent of the periodic testing is also addressed and the recommendations identify procedures which are intended to fully meet the intent of NUREG 0612 and ANSI N14.6 with the least amount of perturbation to the refueling sequence. These recommendations do not involve any equipment changes unless stricter compliance becomes necessary, and then a change to the sling blocks is recommended.

SECTION 5  
CONCLUSIONS

The following conclusions are apparent as a result of this evaluation:

1. The ANSI N14.6 requirements for design, fabrication and quality assurance are generally in agreement with those used for these special lift devices.
2. The ANSI N14.6 criteria for stress limits associated with certain stress design factors for tensile and shear stresses are adequately satisfied.
3. These devices are not in strict compliance only with the ANSI N14.6 requirements for acceptance testing, maintenance and verification of continuing compliance. Recommendations are included to identify actions that should enable these devices to be considered in compliance with the intent of ANSI N14.6.
4. The application of the ANSI N14.6 criteria for stress design factor of 3 and 5 are only for shear and tensile loading conditions. Other loading conditions are to be analyzed to other appropriate criteria.

SECTION 6  
RECOMMENDATIONS

The following recommendations address the areas of ANSI N14.6 which are incompatible with the present lifting devices and which are considered most important in demonstrating the continued reliability of these devices. They consist of suggestions and proposed responses to identify compliance to the NRC and future considerations.

6.1 Review plant operating procedures to include consideration of ANSI N14.6 Sections 5.1.3 through 5.1.8. These sections include requirements for: scheduled periodic testing; special identification and marking; maintenance, repair, testing and use. Westinghouse remarks on addressing these sections are listed in Attachment A, Appendix B, Items 7, 8, and 9.

6.2 A proposed response to the requirement of ANSI N14.6 Section 5.2.1 requiring an initial acceptance load test prior to use equal to 150 percent of the maximum load is as follows:

After site assembly these special lifting devices were subject to a 100 percent load test followed by non-destructive testing of all critical areas. The Unit 2 internals lift rig was subject to a 125 percent load test followed by appropriate non-destructive testing of all critical areas. The new load path items will be tested to 150 percent of the maximum load prior to shipment.

6.3 A proposed response to ANSI N14.6 Section 5.3 which requires, annually, either a 150 percent maximum load test or dimensional, visual and non-destructive testing of major load carrying welds and critical areas follows. (Since the 150 percent load test is very impractical, the approach identified in the following recommendation is to perform a minimum of non-destructive testing.)



a. Reactor Vessel Head Lift Rig

Prior to use and after reassembly of the spreader assembly, lifting lug and upper lifting legs to the upper portion of the lift rig, visually check all welds. Raise the vessel head slightly above its support and hold for 10 minutes. During this time, visually inspect the sling block top lug to top plate weld, side lugs to support pipe welds, and spreader lug to spreader arm weld. If no problems are apparent, continue to lift, monitoring the load cell readout at all times.

b. Reactor Vessel Internals Lift Rig

Prior to use, visually inspect the rig components and welds while on the storage stand for signs of cracks or deformation. Check all bolted joints to ensure that they are tight and secure. After connection to the upper or lower internals, raise the assembly slightly off its support and hold for 10 minutes. During this time, visually inspect the sling block top lug to support plate weld, side lugs to support pipe welds, spacer to leg weld, and brace plate and leg support block to leg welds. If no problems are apparent, continue to lift, monitoring the load cell readout at all times.

The above actions do not include a non-destructive test of these welds because:

- a. Access to the welds for surface examination is difficult. These rigs are in containment and some contamination is present.
- b. All tensile and shear stresses in the welds are well within the allowable stress.

- c. The items that are welded remain assembled and cannot be misused for any other lift other than their intended function.
- d. To perform non-destructive tests would require:
  - (1) Removal of paint around the area to be examined which is contaminated.
  - (2) Performance of either magnetic particle inspection or liquid penetrant inspection and
  - (3) Repainting after testing is completed.
  - (4) Cleanup of contaminated items.

Performing non-destructive tests on these welds every refueling would increase the critical path refueling time.

Dimensional checking is not included since these structures are large (about 14 feet diameter by 30 feet high) and the results of dimensional checking would always be questionable. Other checks on critical load path parts such as pins, are also not included since an examination of these items would require disassembly of the special lift devices.

- 6.4 Recommend that a periodic non-destructive surface examination of critical welds and/or parts be performed once every ten years as part of an inservice inspection outage.
- 6.5 Recommend that the head and internals lift rig sling block be changed to a forged block with welded side lugs to reduce the number of welds and eliminate any concern, if any, of lamellar tearing, should the NRC require yearly surface inspection of welds and plates.



6.6 Recommend that no changes be made to the reactor vessel internals lift rig should the stresses, discussed in Attachment B, be considered excessive by others because:

- a. The design weight used in the stress calculations is based on the weight of the lower internals. The lower internals are only removed when a periodic inservice inspection of the vessel is required (once/10 years).
- b. Prior to removal of the lower internals, all fuel is removed. Thus the concern for handling over fuel is non-existent in this particular case.
- c. Normal use of the rig is for moving the upper internals which weigh less than one-half of the lower internals. The design weight is based on lifting the lower internals. Thus all the stresses could be reduced by more than 50 percent and considered well within the ANSI N14.6 criteria for stress design factors.

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ATTACHMENT A  
to WCAP-10167

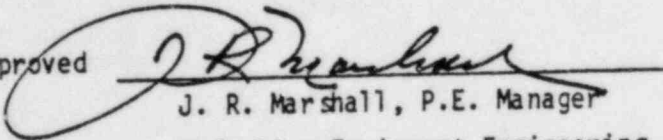
Comparison of ANSI N14.6-1978 Requirements for  
Special Lifting Devices and the Requirements  
for the Reactor Vessel Head Lift Rig, Reactor  
Vessel Internals Lift Rig, Load Cell,  
and Load Cell Linkage  
for

Public Service Electric and Gas Company  
Salem Generating Station  
Units 1 and 2

February 1983

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## ABSTRACT

The requirements used in the original design, fabrication, testing, maintenance and quality assurance were compared to the ANSI N14.6-1978 requirements for the Salem reactor vessel head and internals lift rig, load cell and load cell linkage. A critical items list per ANSI N14.6 section 3.1.2 has been prepared and a tabulation of ANSI N14.6 requirements that are, at present, incompatible with the Salem lifting devices has been prepared.

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A-4	Reactor Vessel Internals Lift Rig, Load Cell and Load Cell Linkage, Critical Items List of Welds per ANSI N14.6-1978	A-12



## REFERENCES

1. Westinghouse Drawing 685J059 - Head Lifting Rig General Assembly
2. Westinghouse Drawing 113E480 - 4 Loop Plants Lifting Rig Internals General Assembly

SECTION 1

PURPOSE

The purpose of this report is to compare the requirements of the special lifting rigs used to lift the reactor vessel head, reactor vessel upper and lower internals with the requirements contained in ANSI N14.6 for special lifting devices.

## SECTION 2 INTRODUCTION

ANSI N14.6-1978-"American National Standard for Special Lifting Devices for Shipping Containers Weighing 10,000 Pounds or More for Nuclear Materials" contains detailed requirements for the design, fabrication, testing, maintenance and quality assurance of special lifting devices. NUREG 0612 "Control of Heavy Load at Nuclear Power Plants", paragraph 5.1.1(4), specifies that special lifting devices should satisfy the guidelines of ANSI N14.6-1978. Subsequently the Nuclear Regulatory Commission (NRC) has requested operating plants to demonstrate compliance with NUREG 0612. To demonstrate compliance with this document, a detailed comparison of the original design, fabrication, testing, maintenance and quality assurance requirements with those of ANSI N14.6 is necessary.

Thus, the ANSI N14.6 document has been reviewed in detail and compared to the requirements used to design and manufacture the reactor vessel head lift rig, the reactor vessel internals lift rig, load cell, load cell linkage and the reactor coolant pump motor lift sling. This comparison is listed in Table 2-1.

### 2.1 BACKGROUND

The reactor vessel head lift rig, the reactor vessel internals lift rig, load cell, and load cell linkage were originally designed and built for the the Salem Generating Station circa 1967-1971. Subsequently, the Unit 2 internals lift rig was transferred to another plant site in 1975-76 and replaced in 1977 with an almost identical design. The actual design criteria is unknown for the lifting devices. It appears that Westinghouse used the design criteria that the resulting stress in the load carrying members, when subjected to the total combined lifting weight, should not exceed one fifth (1/5) of the ultimate strength of the material. These items were not classified as nuclear safety components and requirements for formal documentation of design requirements and stress reports were not applicable. Thus, stress reports and design specifications were not formally documented. Westinghouse defined the

design, fabrication and quality assurance requirements on detailed manufacturing drawings and purchase order documents. Westinghouse also issued field assembly instructions for the reactor vessel head and internals lift rigs which included an initial load test followed by non-destructive surface examination of critical load bearing areas. Additionally, the new Unit 2 internals lift rig was load tested at the manufacturers shop at 125 percent of the maximum weight. Subsequent modifications (presently in progress) to the internals lifting rig include rotolock studs and inserts designed, manufactured, and tested to the requirements of ANSI N14.6.

## 2.2 COMPONENT DESCRIPTION

### 2.2.1 Reactor Vessel Head Lift Rig

The reactor vessel head lift rig<sup>[1]</sup> is a three-legged carbon steel structure, approximately 43 feet high and 14 feet in diameter, weighing approximately 28,000 pounds. It is used to handle the assembled reactor vessel head.

The three vertical legs and control rod drive mechanism (CRDM) platform assembly are permanently attached to the reactor vessel head lifting lugs. The tripod sling assembly is attached to the three vertical legs and is used when installing and removing the reactor vessel head. During plant operations, the sling assembly is removed and the three vertical legs and platform assembly remain attached to the reactor vessel head.

### 2.2.2 Reactor Vessel Internals Lift Rig

The reactor vessel internals lift rig<sup>[2]</sup> is a three-legged carbon and stainless steel structure, approximately 25 feet high and 14 feet in diameter weighing approximately 14,500 pounds. It is used to handle the upper and lower reactor vessel internals packages. It is attached to

the main crane hook for all lifting, lowering and traversing operations. A load cell linkage is connected between the main crane hook and the rig to monitor loads during all operations. When not in use, the rig is stored on the upper internals storage stand.

The original rig design attaches to the internals packages by means of three engaging screws which are screwed into tapped holes in the internals flanges. These screws are manually operated from the manipulator crane walkway using a handling tool which is essentially a long wrench. The screws are normally spring retracted upward and are depressed to engage the tapped holes by the weight of the handling tool. Modifications to these rigs, presently in progress, include installation of an operating platform, integral tools and a change to the type of engagement with the internals to a rotolock stud and insert type. These rotolock studs are manually operated from the new internals lift rig platform using a handling tool which is an integral part of the rig. The studs are normally spring retracted upward and are depressed to engage the inserts. Rotating the mechanism locks it in both positions. Since these rotolock studs are in the process of being incorporated, both types of load carrying items are included in this report.

### 2.2.3 Load Cell and Load Cell Linkage

The load cell is used to monitor the load during lifting and lowering the reactor vessel head or internals to ensure no excessive loadings are occurring. It is installed between the load cell linkage and the lifting device. The load cell is a strain gage (tension) type, rated at 400,000 pounds, built by W. C. Dillon and Co. The load cell linkage is an assembly of pins, plates and bolts which connect the polar crane main hook to the load cell.



TABLE 2-1  
COMPARISON OF THE REQUIREMENTS OF ANSI N14.6 AND  
SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
1 1.1 to 1.3 2  3 3.1 3.1.1 to 3.1.4	<p><u>Scope and Definitions</u> - These sections define the scope of the document and include pertinent definitions of specific items</p> <p><u>Design Designer's Responsibilities</u> - This section contains requirements for preparing a design specification and its' contents, stress reports; repair procedures; limitations on use with respect to environmental conditions; marking and nameplate information; and critical items list.</p>	<p>These sections are definitive, and not requirements.</p> <p>A. No original design specification was written concerning these specific requirements. Subsequent modifications to the internal lift rig meet these requirements. However, assembly and detailed manufacturing drawings and purchasing documents contain the following requirements:</p> <p>(1) Material specification for most of the critical load path items to ASTM, ASME specifications or special listed requirements.</p> <p>(2) All welding, weld procedures and welds to be in accordance with ASME Boiler and Pressure Vessel Code - Section IX.</p> <p>(3) Special non-destructive testing for specific critical load path items to be performed to written and approved procedures in accordance with ASTM or specified requirements</p>

TABLE 2-1 (cont)  
COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
		<p>(4) All coatings to be performed to strict compliance with specified requirements.</p> <p>(5) Letters of compliance for materials and specifications were required for verification with original specifications.</p> <p>B. A stress report was not originally required but has been prepared and is Attachment B.</p> <p>C. Repair procedures were not identified.</p> <p>D. No limitations were identified as to the use of these devices under adverse environments.</p> <p>E. Markings and nameplate information was not addressed.</p> <p>F. Critical item lists have been prepared for each device and are identified as Appendix A to this Attachment A.</p>

TABLE 2-1 (cont)  
 COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
3.2 3.2.1 3.2.6	<p><u>Design Criteria</u>  <u>Stress Design Factors</u> - These sections contain requirements for the use of stress design factors of 3 and 5 for allowable stresses of yield and ultimate respectively for maximum shear and tensile stresses; high strength material stress design factors; special pins; wire rope and slings to meet ANSI B30.9-1971; and drop-weight tests and Charpy impact test requirements</p>	<p>1. The actual design criteria is unknown for the lifting devices. It appears that for the head lifting rig, internals lift rig and load cell, that in most cases the design criteria used was that the resulting stress in the load carrying members, when subjected to the total combined lifting weight, should not exceed one fifth (1/5) of the ultimate strength of the material. A stress report (Attachment B) has been generated which addresses the capability of these rigs to meet the ANSI design stress factors.</p> <p>2. High strength materials are used in some of these devices (mostly for pins, loadcell). Although the fracture toughness was not determined, the material was selected based on it's excellent fracture toughness characteristics. However, the stress design factors of ANSI N14.6 Section 3.2.1 of 3 and 5 were used in the analysis and the resulting stresses are acceptable.</p> <p>3. Where necessary, the weight of pins was considered for handling.</p>

2-6

TABLE 2-1 (cont)  
COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
3.3 3.3.1 to 3.3.8	<p><u>Design Considerations</u> - These sections contain considerations for; materials of construction, lamellar tearing; decontamination effects; remote engagement provisions; equal load distribution; lock devices; position indication of remote actuators; retrieval of device if disengaged; and nameplates.</p>	<p>4. Drop weight and Charpy impact tests were not required nor performed.</p> <p>Decontamination was not specifically addressed. <u>Lamellar tearing</u> was considered in the design of the reactor vessel and internals lift rig sling blocks by requiring non-destructive tests (ultrasonic, magnetic particle and radiograph) of the base material and <u>assembly welds</u>. Even distribution of the load is evident from these designs. Locking plates, pins, etc. are used throughout these special lifting devices. Remote actuation is only used when engaging the internals lift rig with the internals, however, present modifications include positive position indication of engagement. All these items were considered and the designs reflect these requirements.</p>
3.4 3.4.1 to 3.4.6	<p><u>Design Considerations to Minimize Decontamination Efforts in Special Lifting Device Use</u> - These sections contain fabrication, welding, finishes, joint and machining requirements to permit ease in decontamination.</p>	<p>Decontamination was not specifically addressed. However, the design and manufacture included many of these items, i.e. lock devices, pins, etc.</p>



TABLE 2-1 (cont)  
 COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
3.5 3.5.1 to 3.5.10	<u>Coatings</u> - These sections contain provisions for ensuring proper methods are used in coating carbon steel surfaces and for ensuring non-contamination of stainless steel items.	The requirements for coating carbon steel surfaces are contained in a Westinghouse process specification referenced on the assembly and detail drawings. This specification requires a proven procedure, proper cleaning, preparation, application and final inspection of the coating. These requirements meet the intent of 3.5.1 through 3.5.8. No provisions were included in these designs for consideration of decontamination materials or the use of non-contaminating contact materials for use in stainless steel parts.
3.6 3.6.1 to 3.6.3	<u>Lubricants</u> - These sections contain requirements for special lubricants to minimize contamination and degradation of the lubricant and contacted surfaces or water pools	No specific lubrication requirements have been identified. However, neolube is recommended for use with the engaging screws in the internals lift device which are under water and silicone grease for the load cell pins which are out of water.



TABLE 2-1 (cont)  
 COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
4 4.1 4.1.1 to 4.1.12	<p><u>Fabrication</u>  <u>Fabricators Responsibilities</u> - These sections contain specific requirements for proper quality assurance, document control, deviation control, procedure control, material identification and certificate of compliance.</p>	<p>A formal quality assurance program for the manufacturer was not specifically required. However, all the manufacturers welding procedures and non-destructive testing procedures were reviewed by Westinghouse prior to use. All critical load carrying members require letters of compliance for material requirements. Westinghouse performed certain checks and inspections during various steps of manufacturing. Final Westinghouse review includes visual, dimensional, procedural, cleanliness, personnel qualification, etc. and issuance of a quality release to ensure conformance with drawing requirements.</p>
4.2 4.2.1 to 4.2.5	<p><u>Inspectors Responsibilities</u> - These sections contain requirements for a non-supplier inspector.</p>	<p>Westinghouse Quality Assurance personnel performed inprocess and final inspections similar to those identified in these sections. (Also see comments to Section 4.1 above)</p>
4.3 4.3.1 to 4.3.3	<p><u>Fabrication Considerations</u> - These sections contain special requirements for ease in decontamination or control of corrosion.</p>	<p>General good manufacturing processes were followed in the manufacture of these devices. However, the information defined in these sections was not specifically addressed.</p>

TABLE 2-1 (cont)  
 COMPARISON OF REQUIREMENT OF THE ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
5  5.1 5.1.1 to 5.1.8	<p><u>Acceptance Testing Maintenance, and Assurance of Continued Compliance Owner's Responsibilities -</u>            Sections 5.1.1 and 5.1.2 require the owner to verify that the special lifting devices meet the performance criteria of the design specification by reviewing records and witness of testing.</p> <p>Section 5.1.3 requires periodic functional testing</p> <p>Section 5.1.4 requires operating procedure</p>	<p>Both of these rigs were load tested after initial assembly on site followed by non-destructive testing of critical load bearing areas. The new Unit 2 internals lift rig was also load tested to 125% of the design weight. However, the Westinghouse Quality Release may be considered an acceptable alternate to verify that the criteria for letters of compliance for materials and specifications required by the Westinghouse drawings and purchasing documents was satisfied.</p> <p>Maintenance and inspection procedures should be revised to include a visual check of critical welds and parts during lifting to comply with this requirement for functional testing.</p> <p>Operating instructions for the reactor vessel internals lift rig were furnished to the utility and operating procedures were prepared and are used.</p>

TABLE 2-1 (cont)  
COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
	<p>Sections 5.1.5, 5.1.5.1 and 5.1.5.2 require special identification and marking to prevent misuse.</p> <p>Sections 5.1.6, 5.1.7 and 5.1.8 require the owner to provide written documentation on the maintenance, repair, testing and use of these rigs.</p>	<p>It is obvious from their designs that these rigs are special lifting devices and can only be used for their intended purpose. Specific identification of the rig can be made by marking, with stencils, the rig name and rated capacity, preferably on the spreader assembly.</p> <p>Operating instructions and maintenance instructions should be reviewed to assure that they contain the requirements to address maintenance logs, repair and testing history, damage incidents etc.</p>

TABLE 2-1 (cont)  
 COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
5.2 and 5.3 5.2.1 to 5.2.3 and 5.3.1 to 5.3.8	<p><u>Acceptance Testing and Testing to Verify Continuing Compliance</u> - These paragraphs require the rigs to be initially tested at 150% maximum load followed by non-destructive testing of critical load bearing parts and welds and also annual 150% load tests or annual non-destructive tests and examinations; qualification of replacement parts.</p>	<p>The head and internals lift rigs were load tested after field assembly. In addition, the new Unit 2 internals lift rig was load tested to 125% load at initial manufacture. The new load bearing modifications presently in progress will meet these requirements. It is suggested that a check of critical welds and parts be included in the maintenance procedures. Preferably, a visual check during initial lift should be acceptable. Further note that with the use of the load cell for the head and internals, lifting and lowering is monitored at all times. However, the load cell, cannot exceed the rated load by 20% without being inaccurate. This would preclude monitoring of a 150% load test with the present equipment. Replacement parts should be in accordance with the original or equivalent requirements.</p>

TABLE 2-1 (cont)  
 COMPARISON OF THE REQUIREMENT OF ANSI N14.6 AND  
 SALEM SPECIAL LIFT DEVICES

ANSI N14.6 Section	Description of ANSI N14.6 Requirement	Actual Special Lift Device Requirements
5.4 5.4.1 to 5.4.2	<u>Maintenance and Repair</u> - This section requires any maintenance and repair to be performed in accordance with original requirements and no repairs are permitted for bolts, studs and nuts.	Maintenance and repair procedure should contain, as much as possible, requirements that were used in the original fabrication. The critical items list of Appendix A contains the original type of non-destructive testing. The procedure should also define bolts, studs and nuts as non-repairable items.
5.5 5.5.1 to 5.5.2	<u>Non-destructive Testing Procedures, Personnel Qualifications, and Acceptance Criteria</u> - This section requires non-destructive testing to be performed in accordance with the requirements of the ASME Boiler and Pressure Vessel Code	Liquid penetrant, magnetic particle, ultrasonic and radiograph inspections were performed on identified items. These were in accordance with Westinghouse process specifications or as noted on detailed drawings and provide similar results to the requirement of the ASME Code.
6 6.1 6.2 6.3	<u>Special Lifting Devices for Critical Loads</u> - These sections contain special requirements for items handling critical loads.	It is assumed that compliance with NUREG 0612, Section 5.1 has been demonstrated and therefore this section is not applicable to these devices.



### SECTION 3 DISCUSSION

The reactor vessel head and internals lift rigs, load cell and load cell linkage generally meet the intent of the ANSI N14.6 requirements for design and manufacture. However, they are not in strict compliance with the ANSI N14.6 requirements for acceptance testing, maintenance and verification of continuing compliance.

Although no specific design specification was written, the assembly and detailed manufacturing drawings and purchase order documents contain equivalent requirements. A stress report has been prepared for these devices and the design criteria is considered satisfied. These devices, were manufactured under Westinghouse surveillance with identified hold points, procedure review and personnel qualification which adequately meet these related ANSI requirements. Acceptance testing as defined in ANSI N14.6 was not performed, but an initial 100 percent load test was conducted on these special lift devices followed by non-destructive testing of critical areas. In addition, the Unit 2 internals lift rig was 125 percent load tested. The new modifications of load bearing items will comply with ANSI N14.6. The load cell was calibrated in tension at its capacity of 400,000 pounds.

It is anticipated that a 100 percent load test, performed on each device, followed by a visual check of critical welds would be sufficient to demonstrate compliance. This may require modification of Salem operating and maintenance procedures.

SECTION 4  
CONCLUSIONS

The review of the ANSI N14.6 requirements and comparison with the original Westinghouse requirements has shown that these items are generally in agreement for the design, fabrication and quality assurance of the lifting devices. However, the lifting devices are not in strict compliance with the ANSI N14.6 requirements for acceptance testing, maintenance and verification of continuing compliance. These specific requirements that are incompatible with the lifting devices are discussed in Appendix B with suggested actions. Westinghouse's objective was to provide a quality product and this product was designed, fabricated, assembled and inspected in accordance with internal Westinghouse requirements. In general, Westinghouse requirements meet the intent of ANSI N14.6 but not all the specific detailed requirements.

## APPENDIX A

### CRITICAL ITEMS LIST PER ANSI N14.6-1978

#### 1. GENERAL

Section 3.1.2 of ANSI N14.6-1978 specifies that the design specification shall include a critical items list, which identifies critical components and defines their critical characteristics for material, fabrication, non-destructive testing and quality assurance.

"Critical items list" is further defined in ANSI N14.6, Section 2 as:

"critical items list. A list that specifies the items of a special lifting device and their essential characteristics for which specified quality requirements shall apply in the design, fabrication, utilization, and maintenance of the device."

Load carrying members and welds of these special lifting devices are considered to be the critical items.

Tables A-1, A-2, A-3, and A-4, are the critical items list of parts and welds for the reactor vessel head lift rig, the reactor vessel internals lift rig, load cell and load cell linkage, respectively. These tables include the material identification, and the applicable volumetric and surface inspections that were performed in the fabrication of these special lifting devices. In some instances, non-destructive testing was not specified since the material selection and strength result in very low tensile stresses and thus, non-destructive testing was not justified.

The material selection for most critical load path items was made to ASTM, ASME or special material requirements. The material requirements were supplemented by Westinghouse imposed non-destructive testing, and/or special heat treating requirements for all of the critical items. Westinghouse required all welding, welders, and weld procedures

to be in accordance with ASME Boiler and Pressure Vessel Code Section IX for carbon steel welds. Westinghouse required certificates, or letters of compliance that the materials and processes used by the manufacturer were in accordance with the purchase order and drawing requirements. Westinghouse also performed final inspections on these devices and issued quality releases. The new modifications to the internals lift rig pertinent to critical items are also included.

TABLE A-1  
 REACTOR VESSEL HEAD LIFT RIG  
 CRITICAL ITEMS LIST OF PARTS  
 PER ANSI N14.6-1978

Item <sup>(a)</sup>	Description	Material	Non-destructive Testing	
			Material	Finished
1, 3, 6, 10, 15	Pins	ASTM A434 Class BD	Ultrasonic	Magnetic Particle
2	Side Plate	ASTM A514 or USS-T1	Ultrasonic	Magnetic Particle
4, 5	Sling Assembly Link and Lug	ASTM A237 Class A	Ultrasonic	Magnetic Particle
7, 9	Clevis	ASTM A237 Class BD	Ultrasonic	Magnetic Particle
8	Arm	ASTM A306 Grade 70	Ultrasonic	Magnetic Particle
11	Support Lug	ASTM A515 Grade 70	Ultrasonic Magnetic Particle	
12	Ring Girder	ASTM A285 Grade C		
13	Leg	ASTM A36		
14	Clevis Plate	ASTM A515 Grade 70	Ultrasonic Magnetic Particle	

(a) See figure A-1



TABLE A-2  
 REACTOR VESSEL HEAD LIFT RIG  
 CRITICAL ITEMS LIST OF WELDS  
 PER ANSI N14.6-1978

Item <sup>(a)</sup>	Weld Description	Non-destructive Testing	
		Root Pass	Final
4,5	Link Lugs to Link (full penetration)	Visual	Radiograph Magnetic Particle
11,12	Ring Girder to Support Lug (fillet)		Magnetic Particle
13,14	Clevis Plate to Leg (fillet)		Magnetic Particle

(a) See figure A-1.

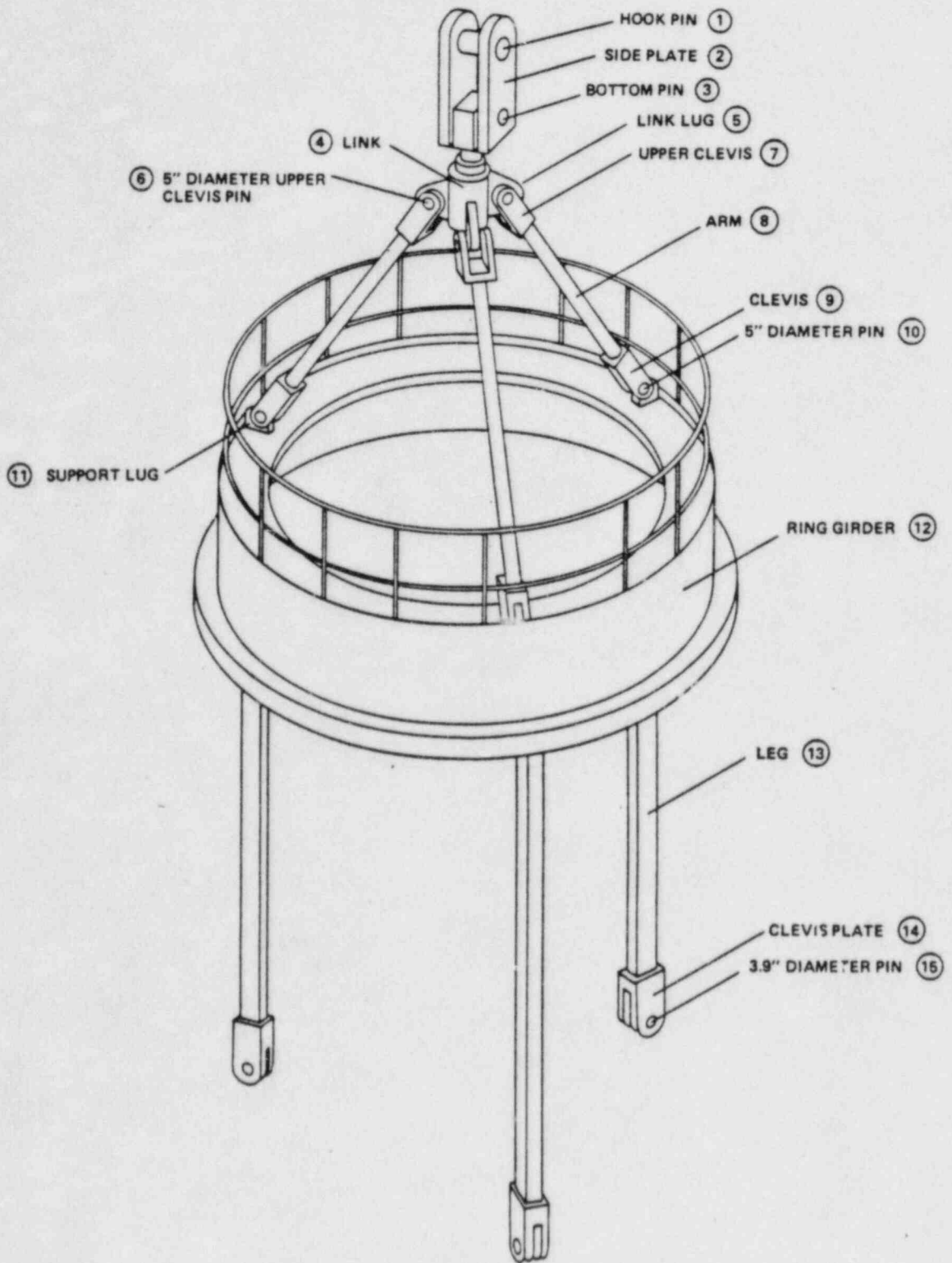


Figure A-1. Reactor Vessel Head Lift Rig

TABLE A-3  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE  
 CRITICAL ITEMS LIST OF PARTS  
 PER ANSI N14.6-1978

Item <sup>(1)</sup>	Description	Material	Non-destructive Testing	
			Material	Finished
1,3,7, 12	7-1/2" Dia. Hook Pin 6" Dia. Adaptor Pin 6" Dia. Removable Pin 4" Dia. Clevis Pin	ASTM A434 Class BD AISI 4340	Ultrasonic	Magnetic Particle
2	Load Cell Linkage Side Plates	ASTM A515 Grade 70	Ultrasonic	
4,6	Adaptors	AISI 4340	Ultrasonic	Magnetic Particle
5	Load Cell	17-4 pH Stain- less Steel, Condition H- 1100	Ultrasonic	Liquid Particle
8a, 10a <sup>(2)</sup>	Top Lug Support Plate	ASTM A515 Grade 70	Magnetic Particle Ultrasonic	
8b,10b <sup>(2)</sup>	Top Lug Support Plate	ASTM 533 Grade B Class 1	Magnetic Particle Ultrasonic	
11a <sup>(2)</sup>	Side Lug	ASTM A515 Grade 70	Magnetic Particle Ultrasonic	
11b <sup>(2)</sup>	Side Lug	ASTM A588 Grade A Class 1	Ultrasonic Magnetic Particle	

(1) See figure A-2

(2) Subscript (a) refers to Unit 1 only, while subscript (b) refers to Unit 2; no subscript means identical to both units.

TABLE A-3 (cont)  
 REACTOR VESSEL INTERNALS LIFT RIG  
 LOAD CELL AND LOAD CELL LINKAGE  
 CRITICAL ITEMS LIST OF PARTS  
 PER ANSI M14.6-1978

Item <sup>(1)</sup>	Description	Material	Non-destructive Testing	
			Material	Finished
13a,15a <sup>(2)</sup>	Clevis	SA508 Class 2	Ultrasonic	Magnetic Particle
13b,15b <sup>(2)</sup>	Clevis	ASTM A668 AISI 4340 Class M	Ultrasonic	Magnetic Particle
14a <sup>(2)</sup>	Sling Leg	AISI 1117 Hot Rolled AISI 1020 Cold Rolled		
14b <sup>(2)</sup>	Sling Leg	ASTM A434 Class BC AISI 4340	Ultrasonic	Magnetic Particle
16	4" Dia. Clevis Bolt	ASTM A434 Class BD AISI 4340	Ultrasonic	
20	Spacer	ASTM A588 Grade A or B		
21,22	Leg Channel Brace Plate	ASTM A36		
23	Leg Support Block	AISI 8620	Ultrasonic Magnetic Particle	

(1) See figure A-2

(2) Subscript (a) refers to Unit 1 only, while subscript (b) refers to Unit 2; no subscript means identical to both units.

TABLE A-3 (cont)  
 REACTOR VESSEL INTERNALS LIFT RIG  
 LOAD CELL AND LOAD CELL LINKAGE  
 CRITICAL ITEMS LIST OF PARTS  
 PER ANSI N14.6-1978

Item <sup>(1)</sup>	Description	Material	Non-destructive Testing	
			Material	Finished
24a <sup>(3)</sup> , 26a <sup>(3)</sup>	Adaptor Guide Sleeve	ASTM A276 Type 304 Condition A	Ultrasonic	Liquid Penetrant
25a <sup>(3)</sup>	Outer Tube	ASTM A312 Type 304 Seamless		
27 <sup>(3)</sup>	Engaging Screw	ASTM A564 Grade 630, Cond. 1100		
28b <sup>(2)</sup>	Load Nut	ASTM A276 Type 304 Cond. A		
29b <sup>(3)</sup>	Tube Housing	ASTM A276 Type 304 Cond. A		
30b <sup>(3)</sup>	Guide Sleeve	ASTM A276 Type 304 Cond. A		

(1) See figure A-2

(2) Subscript (a) refers to Unit 1 only, while subscript (b) refers to Unit 2; no subscript means identical to both units.

(3) These items will be replaced in the forthcoming modifications.



TABLE A-3 (cont)  
 REACTOR VESSEL INTERNALS LIFT RIG  
 LOAD CELL AND LOAD CELL LINKAGE  
 CRITICAL ITEMS LIST OF PARTS  
 PER ANSI N14.6-1978

Item <sup>(1)</sup>	Description	Material	Non-destructive Testing	
			Material	Finished
31a <sup>(4)</sup>	Load Nut	ASTM A276, Type 304 SS Condition A	Ultrasonic	
32 <sup>(4)</sup>	Rod Housing	ASTM A276 Type 304 SS Condition A	Ultrasonic	
33 <sup>(4)</sup>	Guide Sleeve	ASTM A276 Type 304 SS Condition A	Ultrasonic	Liquid Penetrant
34 <sup>(4)</sup>	Rotolock Stud	ASTM A564, Type 630, 17-4 pH, H1100	Ultrasonic	Liquid Penetrant
35 <sup>(4)</sup> , 36 <sup>(4)</sup>	Upper and Lower Internals Inserts	ASTM A637 Grade 688 Type 2 Incone1 X-750	Ultrasonic	Liquid Penetrant

(1) See figure A-2, A-3

(4) These items are the new items to be installed.

WHERE PSE AND PNJ FIGS DIFFER, THE PSE ITEMS ARE DISTINGUISHED BY AN "A" SUFFIX, THE PNJ ITEMS BY A "B" SUFFIX

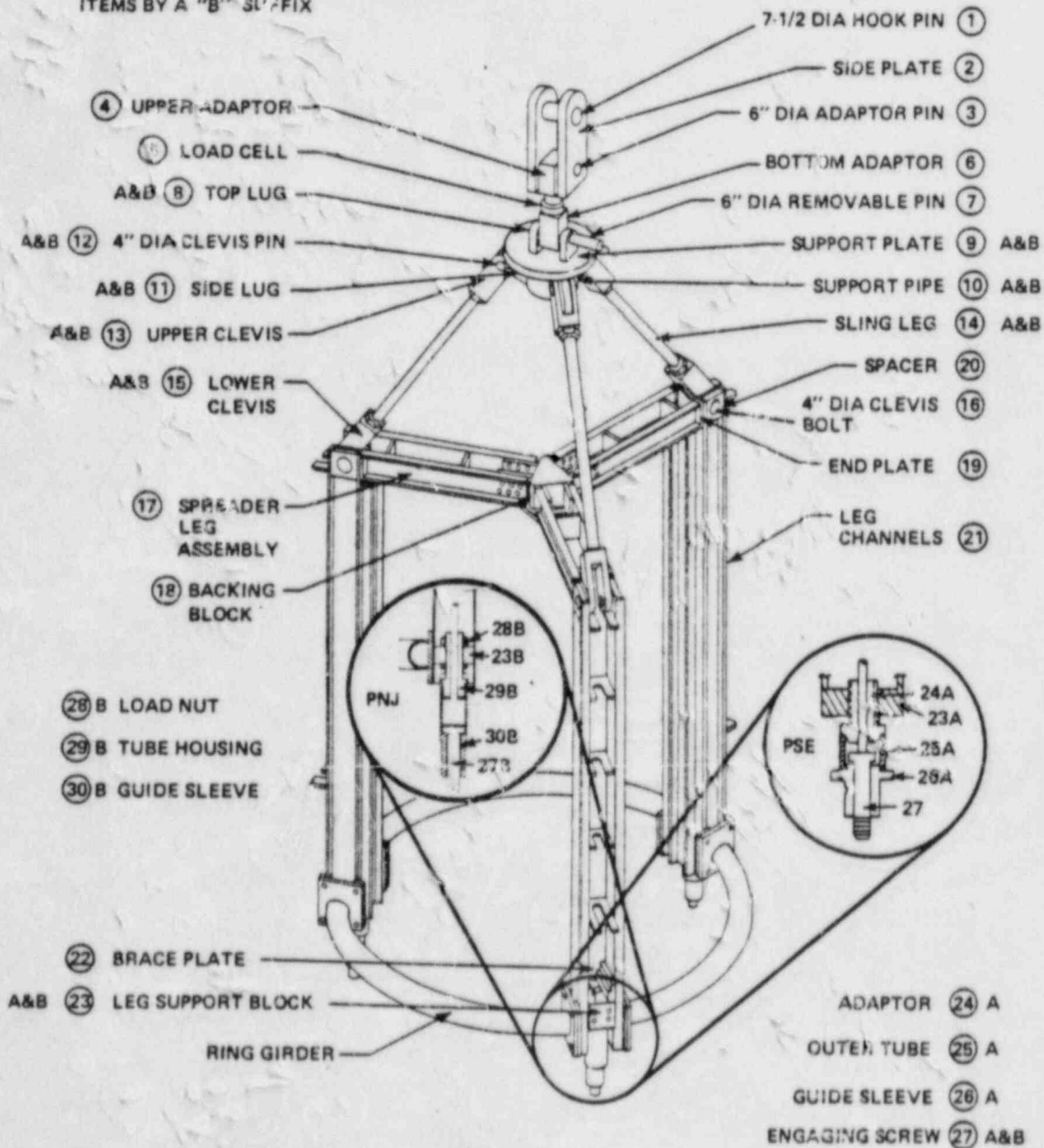
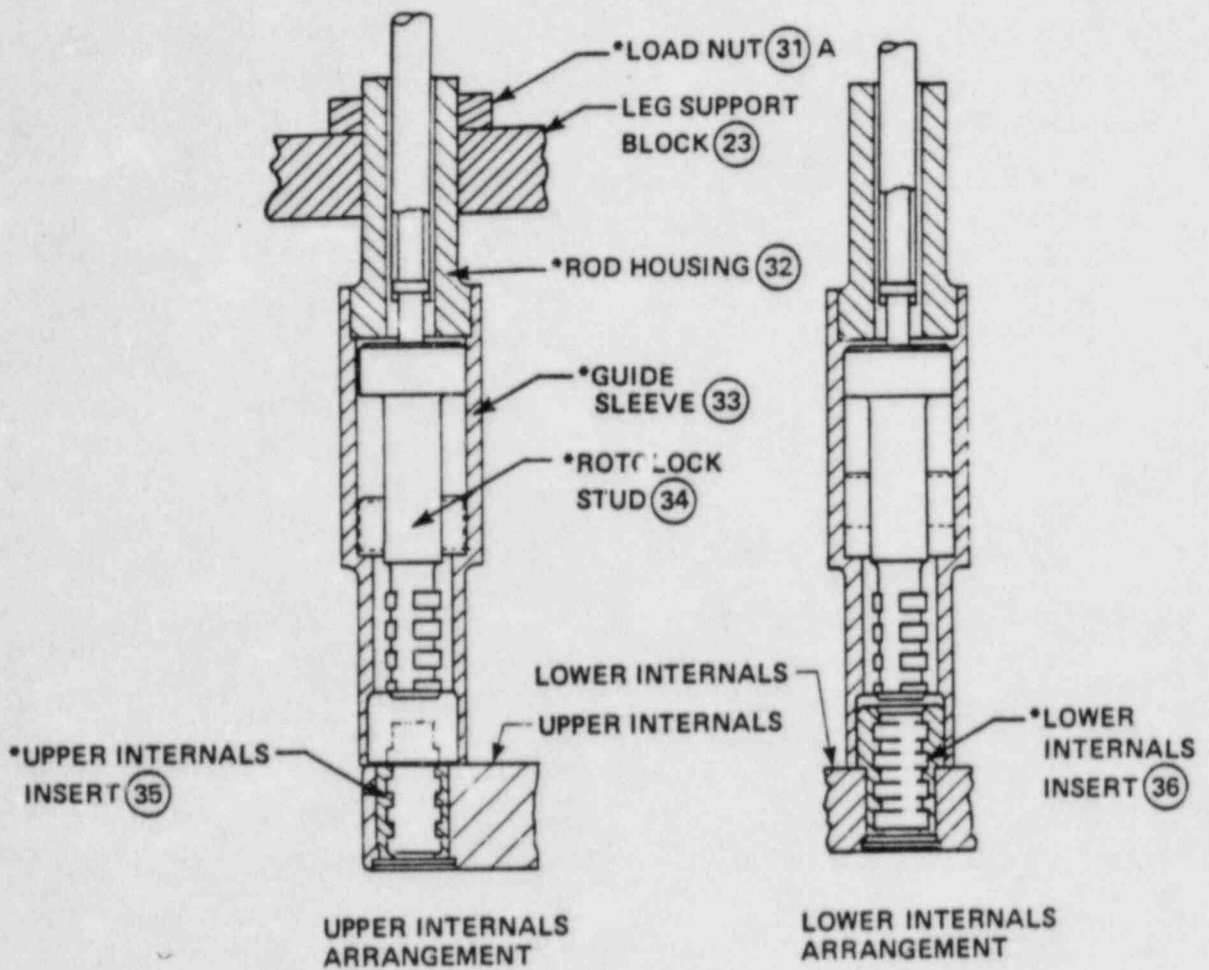


Figure A-2 Reactor Vessel Internals Lift Rig



\*NEW ITEMS  
 LETTER "A" SUFFIX IS FOR UNIT 1 ONLY (PSE)

Figure A-3. RotoLock Studs and Inserts Arrangement

TABLE A-4  
 REACTOR VESSEL INTERNALS LIFT RIG  
 LOAD CELL AND LOAD CELL LINKAGE  
 CRITICAL ITEMS LIST OF WELDS  
 PER ANSI N14.6-1978

Item	Weld Description	Non-destructive Testing	
		Root Pass	Final
8,10	Sling Block Top Lugs to Support Plate (full penetration)	Magnetic Particle	Ultrasonic Magnetic Particle
9,11	Side Lugs to Support Pipe (full penetration)	Magnetic Particle	Radiograph Magnetic Particle
20	Spacer to Leg Weld (Top and Bottom) (full penetration)	Magnetic Particle	Visual Magnetic Particle
24a <sup>(1)</sup> , 25a	Torque Tube Adapter to Outer Tube (full pene- tration)	Liquid Penetrant	Liquid Penetrant
21,22,23	Brace Plate and Leg Support Block to Leg Weld (fillet)	Magnetic Particle	Visual Magnetic Particle

(1) Subscript (a) refers only to Unit 1.

## APPENDIX B

### TABULATION OF ANSI N14.6-1978 REQUIREMENTS INCOMPATIBLE WITH THE SALEM LIFTING DEVICES

#### 1. GENERAL

The comparison of the various ANSI N14.6 requirements and those of these lifting devices has shown that these devices are not in strict compliance with all the ANSI N14.6 requirements. Listed below is a tabulation of those sections of ANSI N14.6 considered most important in demonstrating the continued load handling reliability of these special lifting devices. Associated Westinghouse remarks are also listed and could be used as suggested actions and/or responses to demonstrate compliance to the NRC.

##### 1a. Requirement:

Para. 3.1.4 - requires the designer to indicate permissible repair procedures and acceptance criteria for the repair.

##### 1b. Remarks:

Any repair to these special lifting devices is considered to be in the form of welding. Should pins, bolts or other fasteners need repair, they should be replaced, in lieu of repair, in accordance with the original or equivalent requirements for material and non-destructive testing. Weld repairs should be performed in accordance with the requirements identified in NF-4000 and NF-5000 (Fabrication and Examination) of the ASME Boiler and Pressure Vessel Code, Section III, Division 1 Subsection NF.

##### 2a. Requirement:

Para. 3.2.1.1 - requires the design, when using materials with yield strengths above 80 percent of their ultimate



strengths, to be based on the material's fracture toughness and not the listed design factors.

2b. Remarks:

High strength materials are used in these devices. Although the fracture toughness was not determined, the material was selected based on its excellent fracture toughness characteristics. However, in lieu of a different stress design factor, the stress design factors listed in 3.2.1 of 3 and 5 were used in the analysis and the resulting stresses are considered acceptable.

3a. Requirement:

Para. 3.2.6 requires material for load-bearing members to be subjected to drop-weight or Charpy impact tests.

3b. Remarks:

Fracture toughness requirements were not identified for the material used in these special lifting devices except for the upper and lower clevises of the internals lift rig. The applicable specification for these items is an ASME specification which requires Charpy tests. However, all material selection was based on its excellent fracture toughness characteristics.

4a. Requirement:

Para. 3.3.6 requires an indication that an actuating mechanism is engaged.

4b. Remarks:

The original reactor vessel internals lift rig design employs a long handled tool to engage the rig and the internals. The tool depresses a spring loaded tube and turns the engaging screw into the internals. No specific position indication is identified, except for scribe marks on the tool, and the visual difference in the top of the

spring loaded tube is considered sufficient indication that the internals are engaged. However, the new modifications, presently in progress, include an integral tool with indication of engagement.

5a. Requirement:

Para. 4.1.6 requires a formal quality assurance program for the manufacturer and para. 4.1.7 requires certification and identification of materials.

5b. Remarks:

A formal quality assurance program for the manufacturer was not required for all items. However, the manufacturers welding procedures and non-destructive testing procedures were reviewed by Westinghouse prior to use. All of the critical load carrying members require letters of compliance for material requirements. Westinghouse performed certain checks and inspections during various steps of manufacturing. Final Westinghouse review includes visual, dimensional, procedural, cleanliness, personnel qualification, etc. and issuance of a quality release to ensure conformance with drawing requirements.

6a. Requirement:

Para. 5.1 lists Owner Responsibilities and 5.1.2 requires the owner to verify that the special lifting devices meet the performance criteria of the design specification by records and witness of testing.

6b. Remarks:

There wasn't any design specification for these rigs. Load testing was performed on these lifting devices after field assembly. These were 100 percent load tested and non-destructive testing was conducted on critical welds following the test. In addition, the new Unit 2 internals

lift rig was load tested at assembly to 125 percent of the design weight. However, the Westinghouse Quality Release may be considered an acceptable alternate to verify that the criteria for letters of compliance for materials and specifications required by the Westinghouse drawings and purchasing document were satisfied.

7a. Requirement:

Para. 5.1.3 requires periodic functional testing and a system to indicate continued reliable performance.

7b. Remarks:

Maintenance and inspection procedures should include a visual check of critical welds and parts during lifting to comply with this requirement for functional testing.

8a. Requirement:

Para. 5.1.5, 5.1.5.1 and 5.1.5.2 require special identification and marking to prevent misuse.

8b. Remarks:

It is obvious, from their designs, that these rigs are specific lifting devices and can only be used for their intended purpose and parts are not interchangeable. Specific identification of the rig can be made by marking with stencils, the rig name and rated capacity, preferably on the spreader assembly.

9a. Requirement:

Para. 5.1.6, 5.1.7 and 5.1.8 require the owner to provide written documentation on the maintenance, repair, testing and use of these rigs.

9b. Remarks:

Operating instructions and maintenance instructions should be reviewed to assure that they contain the requirements

to address maintenance logs, repair and testing history, damage incidents and other items mentioned in these paragraphs.

10a. Requirement:

Para 5.2.1 requires the rigs to be initially tested at 150 percent maximum load followed by non-destructive testing of critical load bearing parts and welds.

10b. Remarks

The head lift rig and internals lift rig were all load tested at assembly to 100 percent of the load followed by non-destructive testing of critical welds. In addition, the new Unit 2 internals lift rig was 125 percent load tested at assembly.

11a. Requirement:

Para 5.2.2 requires replacement parts to be individually qualified and tested.

11b. Remarks

Replacement parts, should they be required, should be made of identical (or equivalent) material and inspections as originally required. Only pins, bolt and nuts are considered replacement parts for the reactor vessel head and internal lift rigs.

12a. Requirement:

Para 5.3 requires testing to verify continuing compliance and annual 150 percent load tests or annual non-destructive tests and examinations to be performed.

12b. Remarks

These special lifting devices are used during plant refueling which is approximately once per year. During plant operation these special lifting devices are



inaccessible since they are permanently installed and/or remain in the containment. They cannot be removed from the containment unless they are disassembled and no known purposes exist for disassembly. Load testing to 150 percent of the total weight before each use would require special fixtures and is impractical to perform. It is suggested that a check (visual) of critical welds and parts be conducted at initial lift prior to moving to full lift and movement for these devices. Preferably, a visual check during initial lift should be acceptable. Further note that with the use of the load cell for the head and internals lift rigs, all lifting and lowering is monitored at all times.

## 2. SUMMARY

The requirements for periodic checking and functional load testing appear to be the ANSI N14.6 requirements that are most difficult to demonstrate compliance. It is almost impractical to perform the 150 percent load test prior to each use. It is suggested that the proposal to the NRC include a 100 percent load test to be performed with a minimum of non-destructive testing, (visual-only) in the critical parts and welds.



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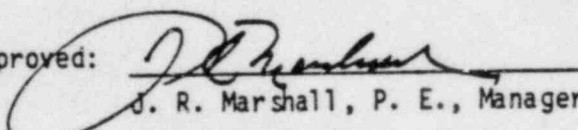
ATTACHMENT B to  
WCAP-10167

Although information contained in this report  
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STRESS REPORT  
REACTOR VESSEL HEAD LIFT RIG,  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE  
FOR  
PUBLIC SERVICE ELECTRIC AND GAS COMPANY  
SALEM GENERATING STATION  
UNITS 1 AND 2

February 1983

L. J. Malandra, P.E.

Approved: 

J. R. Marshall, P. E., Manager  
Refueling Equipment Engineering

## ABSTRACT

A stress analysis of the Salem reactor vessel head and internal lift rigs, load cell and load cell linkage was performed to determine the acceptability of these devices to meet the design requirements of ANSI N14.6.

ACKNOWLEDGMENT

Acknowledgment is hereby made to the following individuals who contributed to the structural analysis presented in this report.

J. W. Richard, P.E.

J. S. Urban

F. Peduzzi

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## SECTION 1 INTRODUCTION

The Nuclear Regulatory Commission (NRC) issued NUREG 0612 "Control of Heavy Load at Nuclear Power Plants"<sup>[1]</sup> in 1980 to address the control of heavy loads to prevent and mitigate the consequences of postulated accidental load drops. NUREG 0612 imposes various training, design, inspection and procedural requirements for assuring safe and reliable operation for the handling of heavy loads. In the containment building, NUREG 0612 requires special lifting devices to meet the requirements of ANSI N14.6-1978 "American National Standard for Special Lifting Devices for Shipping Containers Weighing 10,000 Pounds or More for Nuclear Materials".<sup>[2]</sup> In general, ANSI N14.6 contains detailed requirements for the design, fabrication, testing, maintenance and quality assurance of special lifting devices. In addition, ANSI N14.6 requires that when wire rope or chain is used in the design of a lifting device, the wire rope or chain shall be in conformance with ANSI B30.9-1971 "American National Standard Safety Standard for Slings".<sup>[3]</sup> The NRC has requested operating plants to demonstrate compliance with these requirements.

This report contains the stress analysis performed on the Salem reactor vessel head lift rig, reactor vessel internals lift rig, load cell and load cell linkage to determine the acceptability of these devices to meet these requirements.

### 1.1 BACKGROUND

The reactor vessel head lift rig, the reactor vessel internals lift rig, load cell, and load cell linkage were originally designed and built for the Salem Generating Station circa 1967-1971. Subsequently, the Unit 2 internals lift rig was transferred to another plant site in 1975-76 and replaced in 1977 with an almost identical design. The actual design criteria is unknown for the lifting devices. It appears that Westinghouse used the design criteria that the resulting stress in the load carrying members, when subjected to the total combined lifting

weight, should not exceed one fifth (1/5) of the ultimate strength of the material. These items were not classified as nuclear safety components and requirements for formal documentation of design requirements and stress reports were not applicable. Thus, stress reports and design specifications were not formally documented. Westinghouse defined the design, fabrication and quality assurance requirements on detailed manufacturing drawings and purchase order documents. Westinghouse also issued field assembly instructions for the reactor vessel head and internals lift rigs which included an initial load test followed by non-destructive surface examination of critical load bearing areas. Additionally, the new Unit 2 internals lift rig was load tested at the manufacturers shop at 125 percent of the maximum weight. Subsequent modification (presently in progress) to the internals lifting rig include rotolock studs and inserts designed, manufactured and tested to the requirement of ANSI N14.6.



## SECTION 2 COMPONENT DESCRIPTION

### 2.1 REACTOR VESSEL HEAD LIFT RIG

The reactor vessel head lift rig<sup>[4]</sup> is a three-legged carbon steel structure, approximately 43 feet high and 14 feet in diameter, weighing approximately 28,000 pounds. It is used to handle the assembled reactor vessel head.

The three vertical legs and control rod drive mechanism (CRDM) platform assembly are permanently attached to the reactor vessel head lifting lugs. The tripod sling assembly is attached to the three vertical legs and is used when installing and removing the reactor vessel head. During plant operations, the sling assembly is removed and the three vertical legs and platform assembly remain attached to the reactor vessel head.

### 2.2 REACTOR VESSEL INTERNALS LIFT RIG

The reactor vessel internals lift rig<sup>[5]</sup> is a three-legged carbon and stainless steel structure, approximately 25 feet high and 14 feet in diameter weighing approximately 14,500 pounds. It is used to handle the upper and lower reactor vessel internals packages. It is attached to the main crane hook for all lifting, lowering and traversing operations. A load cell linkage is connected between the main crane hook and the rig to monitor loads during all operations. When not in use, the rig is stored on the upper internals storage stand.

The original rig design attaches to the internals packages by means of three engaging screws which are screwed into tapped holes in the internals flanges. These screws are manually operated from the manipulator crane walkway using a handling tool which is essentially a long wrench. The screws are normally spring retracted upward and are depressed to engage the tapped holes by the weight of the handling tool. Modifications to these rigs, presently in progress, include

installation of an operating platform, integral tools and a change to the type of engagement with the internals to a rotolock stud and insert type. These rotolock studs are manually operated from the new internals lift rig platform using a handling tool which is an integral part of the rig. The studs are normally spring retracted upward and are depressed to engage the inserts. Rotating the mechanism locks it in both positions. Since these rotolock studs are in the process of being incorporated, both types of load carrying items are included in this report.

### 2.3 LOAD CELL AND LOAD CELL LINKAGE

The load cell is used to monitor the load during lifting and lowering the reactor vessel head or internals to ensure no excessive loadings are occurring. It is installed between the load cell linkage and the lifting device. The load cell is a strain gage (tension) type, rated at 400,000 pounds, built by W. C. Dillon and Co. The load cell linkage is an assembly of pins, plates and bolts which connect the polar crane main hook to the load cell.

SECTION 3  
DESIGN BASIS

3.1 DESIGN CRITERIA

NUREG 0612, paragraph 5.1.1(4) states that special lifting devices should satisfy the guidelines of ANSI N14.6. Further, NUREG 0612, 5.1.1(4) states: "In addition, the stress design factor stated in Section 3.2.1.1 of ANSI N14.6 should be based on the combined maximum static and dynamic loads that could be imparted on the handling device based on characteristics of the crane which will be used. This is in lieu of the guideline in Section 3.2.1.1 of ANSI N14.6 which bases the stress design factor on only the weight (static load) of the load and of the intervening components of the special handling device".

It can be inferred from this paragraph that the stress design factors specified in Section 3.2.1.1 of ANSI N14.6 (3 and 5) are not all inclusive. Also, it can be inferred that the specified ANSI N14.6 stress design factors should be increased by an amount based on the crane dynamic characteristics. The dynamic characteristics of the crane would be based on the main hook and associated wire ropes holding the hook. Most main containment cranes use sixteen (16) or more wire ropes to handle the load. Should the crane hook suddenly stop during the lifting or lowering of a load, a shock load could be transmitted to the connected device. Because of the elasticity of the sixteen or more wire ropes, the dynamic factor for a typical containment crane is not much larger than 1.0. The maximum design factor that is recommended by most design texts<sup>[6,7,8]</sup> is a factor of 2 for loads that are suddenly applied. The stress design factors required in Section 3.2.1.1 of ANSI N14.6 are:

3 (weight) < Yield Strength

5 (weight) < Ultimate Strength

The factor of 3 specified, certainly, includes consideration of suddenly applied loads for cases where the dynamic impact factor may be as high

as 2.0. Thus, we feel that the use of the design criteria in ANSI N14.6 satisfies the NUREG requirement.

To provide flexibility on stress design factor, the summary table list the stresses with stress design factors of 1, 3 and 5. Thus, any stress design factor may be easily applied to satisfy any concerns.

### 3.2 DESIGN WEIGHTS

The following design weights were used in the analysis of the lifting devices:

#### 3.2.1 Reactor Vessel Head Lift Rig

The design weight is 345,000 pounds which is the total weight of the reactor vessel head, its attachment and the lift rig.

#### 3.2.2 Reactor Vessel Internals Lift Rig, Load Cell, and Load Cell Linkage

- (a) The design weight for the internals lift rig is 285,000 pounds which is the total weight of the lifting device and the lower internals.
- (b) The design weight is 285,000 pounds for the load cell and load cell linkage.



SECTION 4  
MATERIALS

4.1 MATERIAL DESCRIPTION

The materials and material properties for the reactor vessel head lift rig, the reactor vessel internals lift rig, load cell and load cell linkage are listed in Tables 4-1 and 4-2.



TABLE 4-1  
 REACTOR COOLANT HEAD LIFT RIG MATERIAL AND MATERIAL PROPERTIES

Item(a)	Description	Materials	Yield Strength	Ultimate Strength
			S <sub>y</sub> (ksi)	S <sub>ult</sub> (ksi)
1	7 1/2" Diameter Pin	ASTM A434	100	130
15	3.9" Dia. Bottom Clevis Pin	Class BD	110	140
6,10	5" Dia. Clevis Pins		105	135
3	8" Dia. Pin		100	130
11	Support Lug	ASTM A515 or Gr. 70	38	70
2	Side Plate	ASTM A514 or USS-T1	90	100
4,5	Sling Assembly Link, Lug	ASTM A237 Class A	50	80
13	Leg	ASTM A36	38	70
12	Ring Girder	ASTM A285 GR. C	30	55
7,9	Upper Clevis Clevis	ASTM A237 CT. A	50	80
8	Arm	ASTM A306 Gr. 70	35	70
14	Clevis Plate	ASTM A515 Gr. 70	38	70

(a) See figure 5-1

TABLE 4-2  
 REACTOR VESSEL INTERNALS LIFT RIG, LOAD CELL AND  
 LOAD CELL LINKAGE MATERIAL AND MATERIAL PROPERTIES

Item <sup>(1)</sup>	Description	Material	Yield Strength	Ultimate Strength
			S <sub>y</sub> (ksi)	S <sub>ult</sub> (ksi)
1,3,7, 12	7 1/2" Dia. Hook Pin, 6" Dia. Adaptor Pin, 6" Dia. Removable Pin 4" Dia. Clevis Pin	ASTM A434 AISI 4340 Class BD	120	135
2	Side Plate	ASTM A515 Grade 70	38	70
4,6	Load Cell Adaptors	AISI 4340	120	135
5	Load Cell	17-4 pH s/s Cond. H1100	115	140
8a(2) 10a(2) 11a(2)	Top Lug, Support Plate Side Lug	ASTM A515 Grade 70	38	70
8b(2) 10b(2)	Top Lug Support Plate	ASTM 533 Grade B Class 1	50	80
9	Support Pipe	ASTM A106	30	48
11b(2)	Side Lug	ASTM A588 Grade A Class 1	50	70
13a(2) 15a(2)	Upper Clevis, Lower Clevis	ASME SA 508 Class 2	50	80
13b(2) 15b(2)	Upper Clevis Lower Clevis	ASTM A668 AISI 4340 Class M	110	135

(1) See figures 5-2 and 5-3

(2) Subscript(a) refers to Unit 1 only, while subscript(b) refers to Unit 2;  
 no subscript means identical to both units

TABLE 4-2 (cont)  
 REACTOR VESSEL INTERNALS LIFT RIG, LOAD CELL AND  
 LOAD CELL LINKAGE MATERIAL AND MATERIAL PROPERTIES

Item <sup>(1)</sup>	Description	Material	Yield Strength	Ultimate Strength
			S <sub>y</sub> (ksi)	S <sub>ult</sub> (ksi)
27(3)	Engaging Screw	ASTM A564 Grade 630 Cond. 1100	115	140
28b(2)	Load Nut	ASTM A276 Type 304 Cond. A	30	75
29b(2,3)	Tube Housing	ASTM A276 Type 304 Cond. A	30	75
30b(2,3)	Guide Sleeve	ASTM A276 Type 304 Cond. A	30	75
31a(2,4)	Load Nut	ASTM A276 Type 304 SS Condition A	30	75
32(4)	Rod Housing	ASTM A276 Type 304 SS Cond. A	30	75
33(4)	Guide Sleeve	ASTM A276 Type 304 SS Cond. A	30	75
34(4)	Rotolock Stud	ASTM A564, Type 630, 17-4 Ph, H1100	115	140
35(4), 36(4)	Upper and Lower Internals Inserts	ASTM A637 Grade 688 Type 2 Inconel X-750	115	140

- (1) See figure 5-2 and 5-3.  
 (2) Subscript(a) refers to Unit 1 only, while subscript(b) refers to Unit 2;  
 no subscript means identical to both units.  
 (3) These items will be replaced in the forthcoming modifications.  
 (4) These items are the new items to be installed.

SECTION 5  
SUMMARY OF RESULTS

Tables 5-1 and 5-2, summarize the stresses on each of the parts which make up the reactor vessel head and internals lift rig, load cell, and load cell linkage respectively. All of the tensile and shear stresses meet the design criteria of section 3.2.1.1 of ANSI N14.6, requiring application of stress design factors of three and five with the accompanying allowable stress limits of yield and ultimate strength, respectively. Application of the ANSI N14.6 criteria to structural members subject to compression, bearing, or combined loads result in some stresses exceeding this criteria. However, when using more appropriate criteria, the resulting stresses are considered acceptable.

5.1 DISCUSSION OF RESULTS

5.1.1 Application of ANSI N14.6 Criteria

The design criteria of section 3.2.1.1 of ANSI N14.6, requiring application of stress design factors of three and five with the accompanying allowable stresses, are to be used for evaluating load bearing members of a special lifting device when subjected to loading conditions resulting in shear or tensile stresses. Application of these design load factors to other loading conditions is not addressed in ANSI N14.6. However, these two stress design factors have been used to determine the stresses of the load carrying members when subject to other loading conditions, viz. bearing, bending, buckling. This is an extremely conservative approach and in some cases the resulting stresses exceed the accompanying allowable stress limit.

Structural elements loaded in compression are analyzed by the empirical equations of the ASME and/or AISC rules. Allowable stresses for compression members are based on the estimate that the upper limit of elastic buckling failure is defined by an average column stress equal to one-half of the yield stress. These equations do not determine the limiting stresses of members in buckling but indicate whether or not the calculated stress is or is not within the allowable values. Instead, the ultimate load carrying capability of the member is the determining factor in the structural member's acceptability.



Timoshenko<sup>[9]</sup> notes that the ultimate load for short struts is equal to the material yield point. Calculation of the ultimate load results in this load being larger than the nominal design load and thus, these members are considered acceptable.

#### 5.1.2 Bearing Stresses

For the internals lifting rig, several of the parts do not meet this criteria. However, since they are localized stresses, they can, if necessary, be considered under section 3.2.1.2, which states that the stress design factors of 3.2.1.1 are not intended to apply to situations where high local stresses are relieved by slight yielding. None of the bearing stresses reach the yield stress, and in fact, all of the bearing stresses meet the design criteria of the AISC<sup>[11]</sup> code of 0.9 yield.

#### 5.1.3 Combined Stresses

The combined tensile stress from bending and tension, in the lower sling rod clevis (item 15), of the internals lift rig exceed the section 3.2.1.1 criteria. Bending is not a uniform stress, but is at a maximum at the outermost fiber. Bending contributes to the major portion of the stress shown in the table, and, as a result, the tensile stress without the bending is extremely low and well within the section 3.2.1.1 criteria. The combined stresses also meet the AISC code criteria.

#### 5.1.4 Fillet Weld Stresses

The fillet weld stress at the leg support block to leg weld on the internals lift rig meets the ASME criteria for weld stresses based on base material properties. However, when applying the ANSI N14.6 5W criteria to the nominal stress value, the ASME allowable stress value is exceeded. However, the ANSI N14.6 criteria is satisfied for this weld and thus it is considered acceptable.

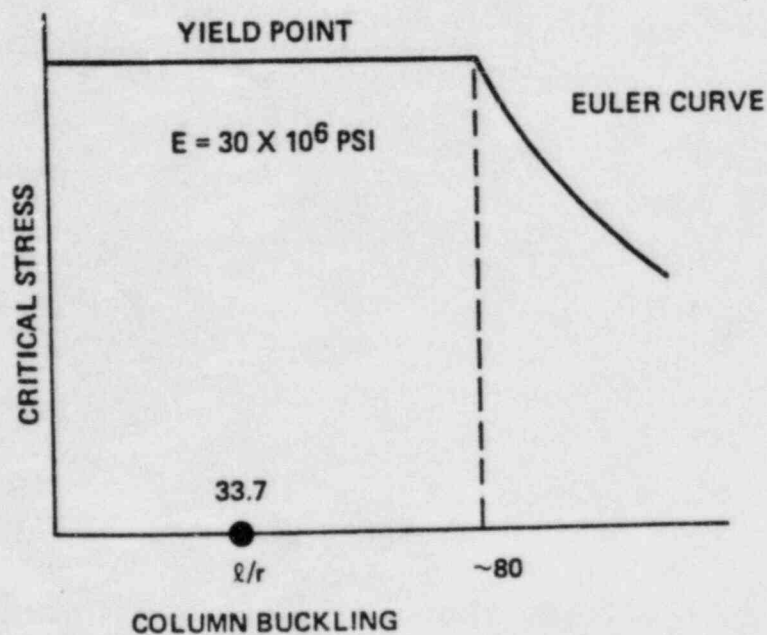


### 5.1.5 Structures Loaded in Compression

The spreader assembly of the reactor vessel internals lift rig, when analyzed for axial compression loadings, does not meet the allowable stresses of the AISC code. However, it does meet the 3W and 5W criteria of the ANSI N14.6 criteria. It is well known that care should be taken when addressing members in compression to ensure elastic stability. Thus, structures loaded in compression are analyzed by the empirical equations of the <sup>[10]</sup> ASME Boiler and Pressure Vessel Code Section III, Appendix XVII or the AISC <sup>[11]</sup> Part 5 rules.

If we were designing a new structure, the material and member size would be changed to ensure these allowable stresses would be satisfied for all loading conditions. However, these calculations are being applied to an existing structure and since these conditions are not satisfied then the ultimate load carrying capability must be determined.

The column under consideration is relatively short ( $\frac{k}{r} = 33.7$ ). Timoshenko <sup>[9]</sup> states that experiments show that short columns buckle when the compressive strength reaches the material yield point. (The horizontal line on the figure below).



Therefore the total stress

$$\sigma = \frac{P}{A}$$

must be less than or equal to the material yield stress.

For the case of the internals lift rig spreader;

$$Q_{\text{Total}} = \frac{P}{A} = 6530 \text{ psi}$$

which is less than the material yield strength ( $S_y$ )

Then to find the ultimate column load, let  $\sigma_{\text{max}} = S_y = 36,000 \text{ psi}$

Then the maximum column load is the ratio of

$$\sigma_{\text{max}}/\sigma_{\text{total}} = \frac{36,000}{6,530} = 5.5$$

Thus the ultimate column load is 5.5 times the nominal value.

The internals lift rig spreader members are considered acceptable for this condition of axial compression.

## 5.2 CONCLUSION

Application of the ANSI N14.6 criteria of (3 and 5) to these special lifting devices results in acceptable stress limits for tensile and shear stresses. Application of this criteria to all structural members subject to other types of loadings tend to result in oversimplified conservatism and with some stresses exceeding the accompanying allowable limits. However, when using the more appropriate criteria for those cases not addressed by the ANSI N14.6 criteria the stresses are within the appropriate allowable limits. Further, these calculations are being applied to an existing structure and if we were designing a new structure, the materials and member sizes would be changed to ensure

these limits would be satisfied for all loading conditions. In conclusion, these special lift devices adequately meet the ANSI N14.6 criteria for tensile and shear stresses and meet other appropriate criteria for loading conditions that result in combined, bearing, and buckling stresses.

TABLE 5-1  
SUMMARY OF RESULTS  
REACTOR VESSEL HEAD LIFT RIG

Item No. <sup>(a)</sup>	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	$3W$	$5W$		
1	7 1/2" Dia. Hook Pin ASTM A434 Class BD	Bending	13.3	39.9	66.5	100	130
		Shear	3.9	11.7	19.5		
		Bearing on Pin	5.8	17.4	29.0		
		Bearing on Side Plate	7.7	23.1	38.5		
2	Side Plate ASTM A514 or USS T1	Tension @ 8" Dia. Hole	6.9	20.7	34.5	90	100
		Shear @ 8" Dia. Hole	6.9	20.7	34.5		
		Bearing @ 7 1/2" Dia. Hole	7.7	23.1	38.5		

(a) See figure 5-1 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)



TABLE 5-1 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL HEAD LIFT RIG

Item (a) No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
3	8" Dia. Bottom Pin ASTM A434 Class BD	Bending	10.9	32.7	54.5	100	130
		Shear	3.4	10.2	17.0		
		Bearing on Lug	5.4	16.2	27.0		
		Bearing on Side Plate	7.2	21.6	36.0		
4	Sling Assembly Link ASTM A237 Class A	Tension @ Hole	5.0	15.0	25.0	50	80
		Shear @ Hole	5.0	15.0	25.0		
		Bearing @ Hole	5.4	16.2	27.0		
		Tension @ Shank	6.6	19.8	33.0		

- (a) See figure 5-1 for location of item numbers and section  
 (b) W is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)

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5-7



TABLE 5-1 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL HEAD LIFT RIG

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
5	Link Lug ASTM 237 Class A	Tension @ Hole	4.0	12.0	20.0	50	80
		Shear Tearout @ Hole	4.0	12.0	20.0		
		Bearing @ Hole	6.3	18.9	31.5		
		Maximum Tension @ Root of Lug	3.7	11.1	18.5		
		Vertical Shear @ Root of Lug	1.4	4.2	7.0		
6	5" Dia. Clevis Pin ASTM A434 Class BD	Bending	10.5	31.5	52.5	105	135
		Shear	3.2	9.6	16.0		
		Bearing on Clevis	5.3	15.9	26.5		
		Bearing on Lug	6.3	18.9	31.5		

- (a) See figure 5-1 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-1 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL HEAD LIFT RIG

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
7	Upper Clevis ASTM A237 Class A	Tension @ Hole	3.2	9.6	16.0	50	80
		Shear @ Hole	3.2	9.6	16.0		
		Bearing @ Hole	5.3	15.9	26.5		
		Thread Shear	2.3	6.9	11.5		
8	Arm ASTM A306 Gr. 70	Thread Tension	7.1	21.3	35.5	35	70
		Thread Shear	2.3	6.9	11.5		
9	Bottom Clevis ASTM A237 Class A	Stresses are the same as Item 7.	Same as Item 7	Same as Item 7	Same as Item 7	50	80

(a) See figure 5-1 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-1 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL HEAD LIFT RIG

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
$W^{(b)}$	3W		5W				
10	5" Dia. Bottom Clevis Pin ASTM A434 Class BD	Stresses are the same as Item 6	Same as Item 6	Same as Item 6	Same as Item 6	105	135
11	Support Lug ASTM A515 Gr. 70	Tension @ Hole Shear @ Hole Bearing @ Hole	4.0 4.0 6.5	12.0 12.0 19.5	20.0 20.0 32.5	38	70
12	Ring Girder ASTM A285 Gr. C	Total Shear Maximum Bending Maximum Tensile Stress Ring Girder to Support Weld	3.2 2.6 4.8 3.2	9.6 7.8 14.4 9.6	16.0 13.0 24.0 16.0	30   18 <sup>(e)</sup>	55

(a) See figure 5-1 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(e) Stress limit for fillet weld from ASME Boiler & Pressure Vessel Code, Section III, Division 1 - Subsection NF 1980 Edition, Table NF - 3292.1-1 page 50

TABLE 5-1 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL HEAD LIFT RIG

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{uit}^{(d)}$
			$w^{(b)}$	3W	5W		
13	Leg ASTM A36	Tension	9.7	29.1	48.5	38	70
14	Clevis Plate ASTM A515 Gr. 70	Weld	5.6	16.8	28.0	18 <sup>(e)</sup> 38	70
		Tension	3.2	9.6	16.0		
		Shear	3.2	9.6	16.0		
		Bearing	7.3	21.9	36.5		
15	3.9" Diameter Pin ASTM A434 Class BD	Bending	18.9	56.7	94.5	110	140
		Shear	4.7	14.1	23.5		
		Bearing on Lug	7.3	21.9	36.5		
		Bearing on Clevis	7.0	21.0	35.0		

(a) See figure 5-1 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d) S is the ultimate strengt



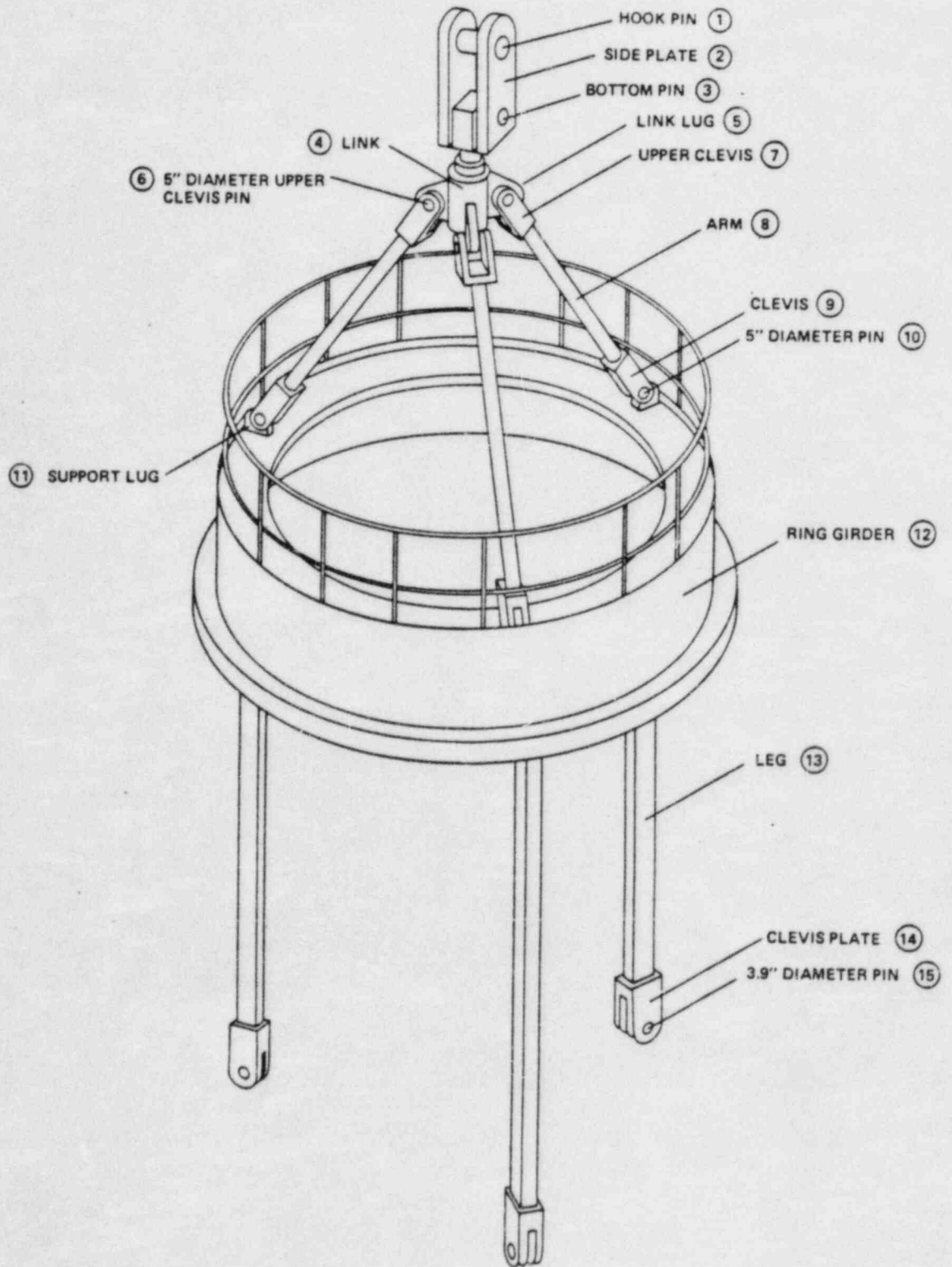


Figure 5-1. Reactor Vessel Head Lift Rig



TABLE 5-2  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item No. (a)	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$W$ (b)	$3W$	$5W$		
1	Hook Pin ASTM A434 AISI 4340	Shear	3.6	10.8	18.0	120	135
		Bearing (Hook)	5.2	15.6	26.0		
		Bearing (Side Plate)	10.9	32.7	54.5		
		Bending	12.1	36.3	60.5		
2	Side Plate ASTM A515 Gr. 70	Tension @ 7 1/2" Dia. hole	10.9	32.7	54.5	38	70
		Bearing @ 6" Dia. hole	13.6	40.8	68.0		
		Shear Tear-out	14.0	42.0	70.0		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)  
 (e) Actual certified mechanical properties

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
3	Adaptor Pin ASTM A304 AISI 4340	Shear	5.7	17.1	28.5	120	135
		Bearing (Adaptor)	6.5	19.5	32.0		
		Bearing (Side Plate)	13.6	40.8	68.0		
		Bending	23.7	71.1	118.5		
4	Upper Adaptor AISI 4340	Tension @ Pin Hole	14.8	44.4	74.0	120	135
		Bearing @ Pin Hole	6.4	19.2	32.0		
		Tension @ Thread Relief	5.7	17.1	28.5		
		Thread Shear	13.7	41.1	68.5		
		Shear Tear-out	6.8	20.4	34.0		

(a) See figure 5-2 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	$3W$	$5W$		
5	Load Cell 17-4-pH s's Cond H-1100	Tension @ Thread	23.9	71.7	119.5	115	140
		Thread Shear	13.7	41.1	68.5		
6	Lower Adaptor AISI 4340	Tension @ Pin Hole	15.9	47.7	79.5	120	135
		Bearing @ Pin Hole	8.5	25.5	42.5		
		Tension @ Thread Relief	7.4	22.2	37.0		
		Thread Shear	13.7	41.1	68.5		
		Shear Tear-out	8.7	26.1	43.5		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
7	Removable Pin ASTM A304 AISI 4340	Shear	5.0	15.0	25.0	120	135
		Bearing on Adaptor	7.6	22.8	38.0		
		Bearing on Top Lug	7.9	23.7	39.5		
		Bending	21.0	63.0	105.0		
8 <sub>a</sub>	Top Lug ASTM A515 Gr. 70	Tension @ Hole	8.0	24.0	40.0	8 <sub>a</sub> 38	70
		Bearing @ Hole	7.9	23.7	39.5		
		Tension @ Weld	4.0	12.0	20.0		
		Shear Tear-out	8.0	24.0	40.0		
8 <sub>b</sub>	ASTM A533 Gr. B, Cl 1					8 <sub>b</sub> 50	80

(a) See figure 5-2 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
9	Support Pipe ASTM A106	Tension	3.1	9.3	15.5	30	48
10 <sub>a</sub>	Support Plate ASTM A515 Gr. 70	Tension	.7	2.1	3.5	38	70

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)



TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
10 <sub>b</sub>	ASTM A533 Gr. B, Cl. 1	Tension	1.1	3.3	5.5	50	80
11 <sub>a</sub>	Side Lug ASTM A515 Gr. 70	Tension @ Hole	5.1	15.3	25.5	38	70
		Combined Stress @ Weld	6.5	19.5	32.5		
		Shear @ Weld	2.0	6.0	10.0		
		Bearing @ Hole	7.5	22.5	37.5		
		Shear Tear-out	4.5	13.5	22.5		

- (a) See figure 5-2 for location of item numbers and section  
 (b) W is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item No.	Part Name And Material	Calculated Stresses (ksi)			Material Allowable (ksi)		
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
11 <sub>b</sub>	Side Lug ASTM A588 Gr. A, Cl. 1	Tension @ Hole	5.2	15.6	26.0	50	80
		Combined Stress @ weld	6.7	20.1	33.5		
		Shear @ Weld	2.0	6.0	10.0		
		Bearing @ Hole	7.7	23.1	38.5		
		Shear Tear-out	4.5	13.5	22.5		
12	Clevis Pin ASTM A434 AISI 4340	Shear	4.7	14.1	23.5	120	135
		Bearing On Side Lug	7.5	22.5	37.5		
		Bearing on Upper Clevis	7.5	22.5	37.5		
		Bending	19.2	57.6	96.0		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item No. (a)	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$W$ (b)	$3W$	$5W$		
13 <sub>a</sub>	Upper Clevis SA-508 Cl. 2	Tension @ Pin Hole	7.7	23.1	38.5	13 <sub>a</sub> 50	80
13 <sub>b</sub>	ASTM A668 AISI 4340 Cl. M	Bearing on Pin	7.5	22.5	37.5	13 <sub>b</sub> 110	135
		Thread Shear	3.3	9.9	16.5		
		Tear-out Shear	7.7	23.1	38.5		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	$3W$	$5W$		
14 <sub>a</sub>	Sling Leg AISI 1117 Hot Rolled or 1018 Cold Rolled	Thread Shear	3.2	9.6	16.0	45	69
		Tension @ Thread	10.7	32.1	53.5	or 32	60
14 <sub>b</sub>	Sling Leg ASTM A434 Cl. BC AISI 4340	Thread Shear	3.2	9.6	16.0	85	110
		Tension @ Thread	11.6	34.8	58.0		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	$3W$	$5W$		
15 <sub>a</sub>	Lower Clevis SA-508 Cl. 2 ASTM A668	Thread Shear Combined Stress	3.3 30.6 0.87 <sup>(f)</sup>	9.9 91.8	16.5 153.0	15 <sub>a</sub> 50 15 <sub>b</sub> 110 Ratio <1.0 <sup>(f)</sup>	80 135
15 <sub>b</sub>	AISI 4340 Cl. M						

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(e) Actual certified mechanical properties

(f) Stress ratio Limit for combined bending and tension from AISC manual Chapter 5, Sect. 1.6.2



TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
16	Clevis Bolt ASTM A434 Cl. BC AISI 4340	Shear	5.6	16.8	28.0	85	110
		Bending	7.8	23.4	34.0		
		Bearing	14.9	44.7	74.5		
17	Spreader Leg ASTM A36	Compression	6.5	19.5	32.5	36	58 $F_a = 18.9^{(g)}$

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(g)  $F_a$  is the compressive buckling strength of the material (ksi)

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$w$ (b)	3W	5W		
18	Backing Block ASTM A276 Type 304 Cond. A	Bearing on Spreader Leg	11.7	35.1	58.5	30	75
		Compression in Block	7.9	23.7	39.5		
19	End Plate ASTM A36	Bearing on Channel	6.5	19.5	32.5	36	58

(a) See figure 5-2 for location of item numbers and section

(b)  $w$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
20	Spacer ASTM A588 Gr. A and B	Bearing	14.9	44.7	74.5	50 61.5 <sup>(e)</sup> Ratio <1.0 <sup>(f)</sup>	70 88.6 <sup>(e)</sup>
		Combined Stress @ Hole	25.3	75.9	126.5		
			0.995				
		Shear in Weld	1.8	5.4	9.0		
21	Leg Channel ASTM A36	Bearing	14.9	44.7	74.0	36	58
		Tension @ Cross-section	8.7	26.1	43.5		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(e) Actual certified mechanical properties

(f) Stress ratio Limit for combined bending and tension from AISC manual Chapter 5, Sect. 1.6.2

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)			Material Allowable (ksi)		
		Designation	Value		$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>	
22,23 <sub>a</sub>	Leg Support Block AISI 8620 Brace Plate ASTM A36	Shear (weld)	4.8	14.4	24.0	18 <sup>(h)</sup>	58
		Bearing on Adaptor Nut	8.8	26.4	44.0	36	

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(h) Stress limit for fillet weld from ASME Boiler & Pressure Vessel Code, Section III, Division 1-  
 Subsection NF 1980 Edition, Table NF-3292.1-1. page 50

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item No. (a)	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$w$ (b)	3W	5W		
24 <sub>a</sub> (i)	Adaptor ASTM A276 Type 304 Cond. A	Thread Shear	2.3	6.9	11.5	30	75
		Tension @ Thd Relief	8.3	24.9	41.5		
25 <sub>a</sub> (i)	Outer Tube ASTM A312 Type 304	Thread Shear	5.7	17.1	28.5	30	70
		Tension @ Thd Relief	9.6	28.8	48.0		

(a) See figure 5-2 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(i) These items will be replaced in the forthcoming modifications



TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
26 <sub>a</sub> <sup>(1)</sup>	Guide Sleeve ASTM A276 Type 304 Cond. A	Bearing On Engaging Screw	17.0	51.0	85.0	30	75
		Thread Shear	5.7	17.1	28.5		
		Compression	10.1	30.3	50.5		
27 <sub>a</sub> <sup>(1)</sup>	Engaging Screw ASTM A564 Grade 630 Cond. 1100	Bearing on Guide Sleeve	17.0	51.0	85.0	115	140
		Tension @ Thread Relief	12.6	37.8	63.0		
		Thread Shear	6.6	19.8	33.0		
27 <sub>b</sub> <sup>(1)</sup>		Bearing on Guide Tube	12.4	37.2	62.0		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(e) Actual certified mechanical properties

(1) These items will be replaced in the forthcoming modifications

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
28 <sub>b</sub>	Load Nut ASTM A276 Type 304 Cond. A	Bearing	8.8	26.4	44.0	30	75
		Thread Shear	4.7	14.1	23.5		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)  
 (i) These items will be replaced in the forthcoming modifications

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y^{(c)}$	$S_{ult}^{(d)}$
			$W^{(b)}$	3W	5W		
29 <sub>b</sub> (i)	Tube Housing ASTM A276 Type 304 Cond. A	Thread Shear	4.1	12.3	20.5	30	75
		Tension	9.0	27.0	45.0		
30 <sub>b</sub> (i)	Guide Sleeve ASTM A276 Type 304 Cond. A	Thread Shear	4.1	12.3	20.5	30	75
		Tension	10.3	30.9	51.5		
		Bearing	12.4	36.8	62.0		

(a) See figure 5-2 for location of item numbers and section

(b) W is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(i) These items will be replaced in the forthcoming modifications

TABLE 5-2 (cont)  
SUMMARY OF RESULTS  
REACTOR VESSEL INTERNALS LIFT RIG,  
LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
31 <sub>a</sub> (j)	Load Nut ASTM A276 Type 304	Bearing to Mounting Block	8.7	26.1	43.5	30	75
		Thread Shear	4.5	13.5	22.5		
32 <sub>a</sub> (j)	Rod Housing ASTM A276 Type 304	Tension @ Thread Relief	9.1	27.3	45.5	30	75
		Thread Shear on Upper Threads	4.5	13.5	22.5		
		Lower Threads Shear	4.5	13.5	22.5		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)  
 (j) These items are the new items to be installed

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item <sup>(a)</sup> No.	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ <sup>(c)</sup>	$S_{ult}$ <sup>(d)</sup>
			$W$ <sup>(b)</sup>	3W	5W		
32 <sub>b</sub> <sup>(j)</sup>	Rod Housing ASTM A276 Type 304	Tension @ Thd Relief	7.8	23.4	39.0	30	75
		Thread Shear in Upper Threads	4.2	12.6	21.0		
		Lower Threads - Shear	4.5	13.5	22.5		
33 <sup>(j)</sup>	Guide Sleeve ASTM A276 Type 304 SST	Thread Shear	4.5	13.5	22.5	30	75
		Tension @ Thread Relief	9.7	29.1	48.5		
		Bearing to Stud	9.9	29.7	49.5		

- (a) See figure 5-2 for location of item numbers and section  
 (b)  $W$  is the total static weight of the component and the lifting device  
 (c)  $S_y$  is the yield strength of the material (ksi)  
 (d)  $S_{ult}$  is the ultimate strength of the material (ksi)  
 (j) These items are the new items to be installed



TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item No. (a)	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$W$ (b)	$3W$	$5W$		
34(j)	Rotolock Stud ASTM A564 Type 630 17-4 pH H100	Tensile Stress @ Cross- Section	27.7	83.1	138.5	115	140
		Shear Stress on Land Root	9.7	29.1	48.5		
		Bearing on Land Surfaces	50.3	150.9	251.5		
		Bearing on Guide Sleeve	25.9	77.7	129.5		
		Bending in Lands	9.9	29.7	49.5		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

(j) These items are the new items to be installed

TABLE 5-2 (cont)  
 SUMMARY OF RESULTS  
 REACTOR VESSEL INTERNALS LIFT RIG,  
 LOAD CELL AND LOAD CELL LINKAGE

Item No. (a)	Part Name And Material	Calculated Stresses (ksi)				Material Allowable (ksi)	
		Designation	Value			$S_y$ (c)	$S_{ult}$ (d)
			$W$ (b)	$3W$	$5W$		
35	Upper Internal Insert ASTM A637 Gr 688 Type 2	Bending In Lands	13.7	41.1	68.5	115	170
		Shear Stress on Land Root	5.6	16.8	28.0		
		Bearing on Land Surfaces	41.5	124.5	207.5		
		Thread Shear	2.3	6.9	11.5		
36	Lower Internal Insert ASTM A637 Gr 688 Type 2	Bending Stress	19.8	59.4	99.0	115	170
		Shear Stress on Land Root	8.1	24.3	40.5		
		Bearing on Land Surfaces	59.9	179.7	299.5		
		Thread Shear	6.9	20.7	34.5		

(a) See figure 5-2 for location of item numbers and section

(b)  $W$  is the total static weight of the component and the lifting device

(c)  $S_y$  is the yield strength of the material (ksi)

(d)  $S_{ult}$  is the ultimate strength of the material (ksi)

WHERE PSE AND PNJ RIGS DIFFER, THE PSE ITEMS ARE DISTINGUISHED BY AN "A" SUFFIX, THE PNJ ITEMS BY A "B" SUFFIX

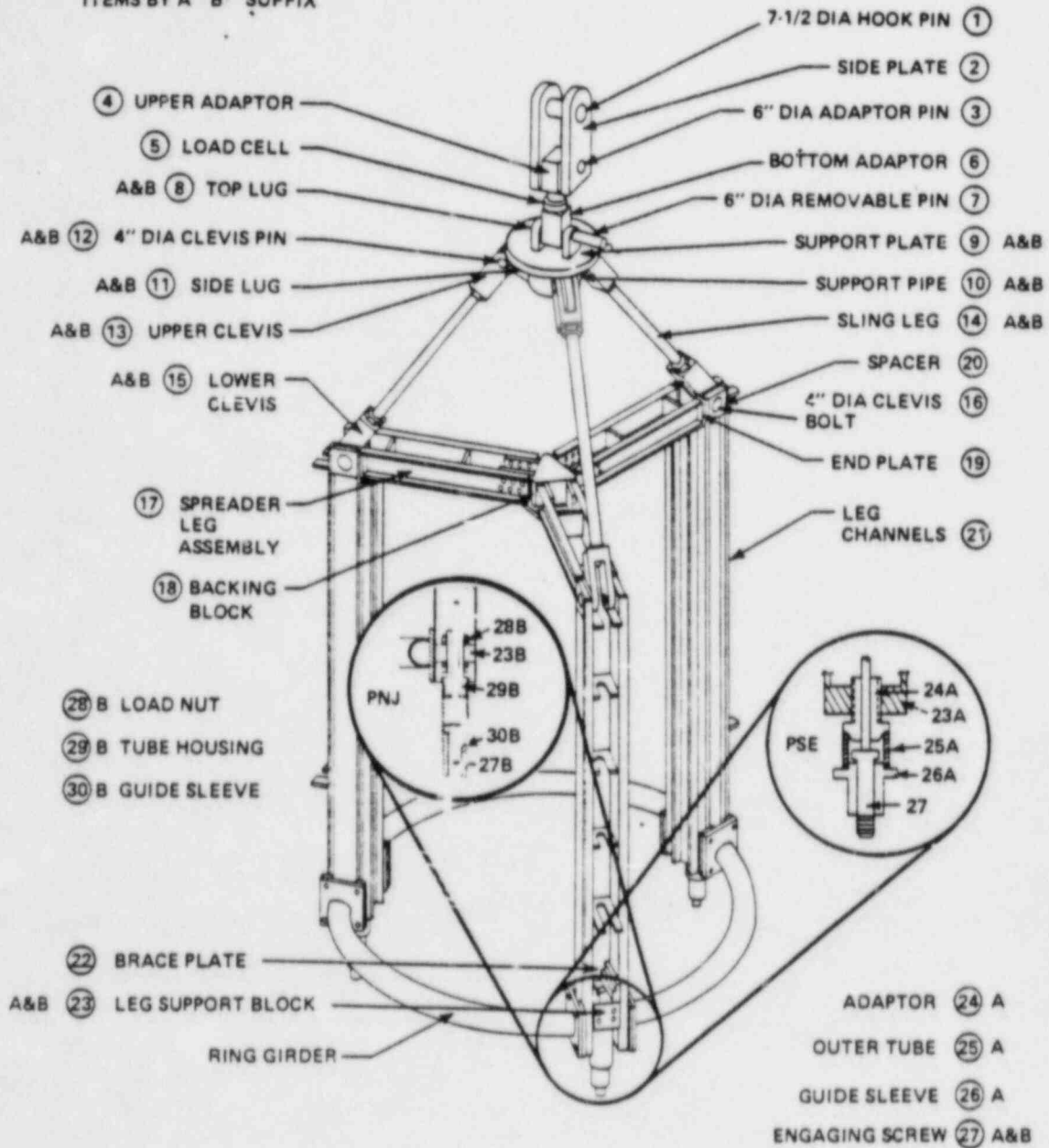
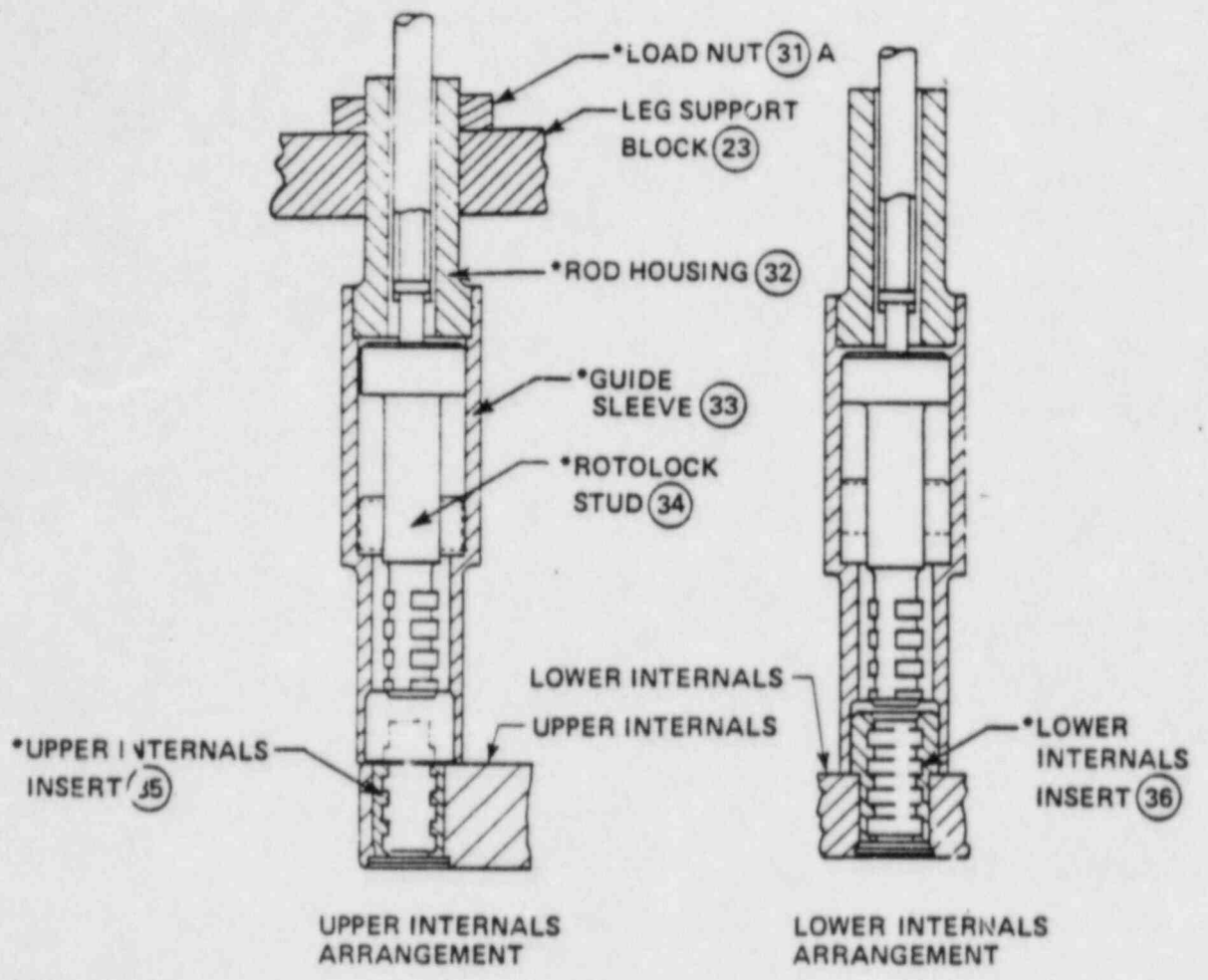


Figure 5-2. Reactor Vessel Internals Lift Rig, Load Cell, and Linkage



\*NEW ITEMS  
 LETTER "A" SUFFIX IS FOR UNIT 1 ONLY (PSE)


Figure 5-3. RotoLock Studs and Inserts Arrangement

APPENDIX A  
DETAILED STRESS ANALYSIS - REACTOR VESSEL HEAD LIFT RIG

This appendix provides the detailed stress analysis for the Salem reactor vessel head lift rig in accordance with the requirements of ANSI N14.6. Acceptance criteria used in evaluating the calculated stresses are based on the material properties given in section 4.

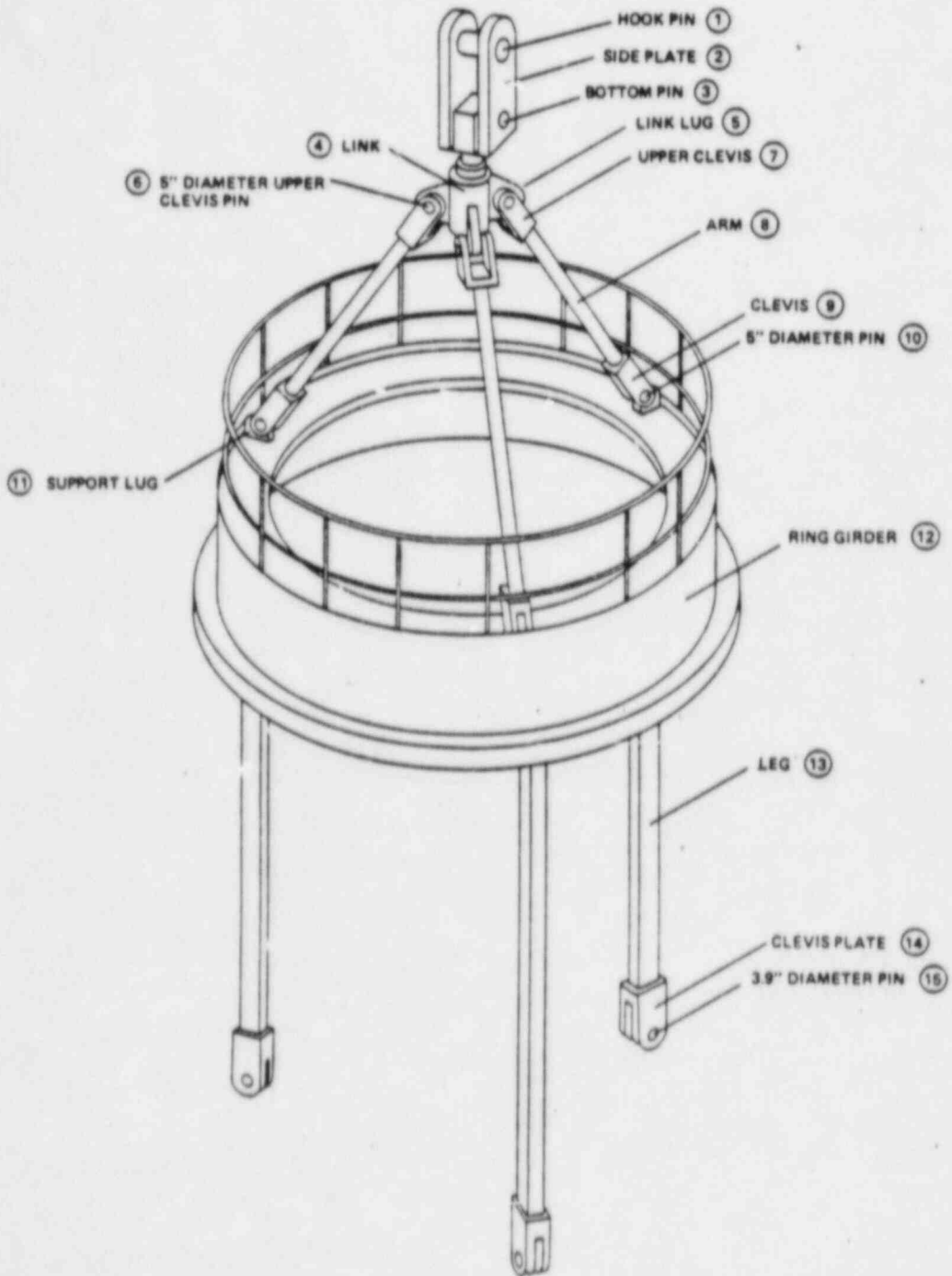


SKETCH SHEET  
WESTINGHOUSE FORM 54202

S.O. PTNP-188	PROJECT Salem Units I and II	PAGE 1 of 33	
TITLE R.V. Head Lift Rig Assembly		CALCULATIONS NO. PDC-	
AUTHOR & DATE F. Peduzzi <i>F.C. Peduzzi</i> 1/83	CHECKED BY & DATE J. Richard <i>J. W. Richard</i> 1/83		
PURPOSE AND RESULTS:			
<ol style="list-style-type: none"> <li>1. The purpose of this analysis is to determine the acceptability of this rig to the requirements of ANSI N14.6.</li> <li>2. The results show that all stresses are within the allowable stresses.</li> </ol>			
 <p><i>J. W. Richard</i></p>			
Original Issue			
REVISION NO.	DATE	DESCRIPTION	BY
RESULTING REPORTS, LETTERS OR MEMORANDA:			

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R. V. Head Lift Rig						2 of 33	
PROJECT	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE	
PSE/PNJ	G. C. Peduzzi	1/73					
S.O.	CALC. NO.	FILE NO.	GROUP		CHE		
PTNP-188							



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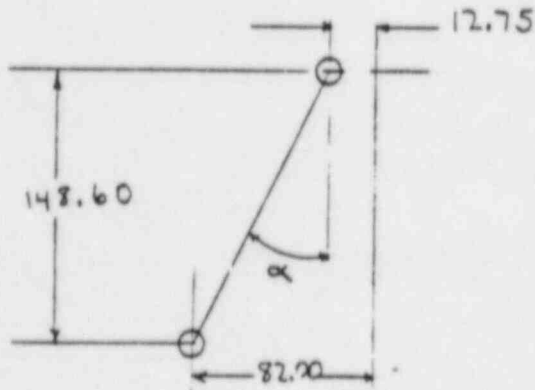
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TITLE R. V. Head Lift Rig		PAGE 3 OF 33	
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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

DESIGN WEIGHT

310,000 lbs From ASSY DWG (625J058)  
 28,000 lbs Rig weight  
7,000 lbs Contingencies  
 345,000 lbs

SLING LEG ANGLE



$$\tan \alpha = \frac{82.00 - 12.75}{148.60} = \frac{69.25}{148.60} = .4660$$

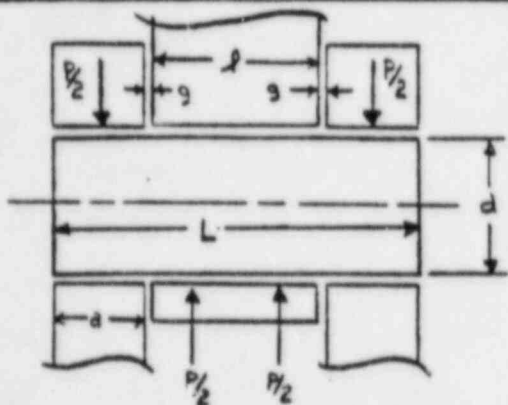
$$\alpha = 25^\circ$$

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7.5" DIA PIN 1  
 MAT'L: ASTM A 434 CLASS BD



P = force acting on assembly, lb.  
 d = diameter of pin, in.  
 l = length of bearing surface of center body, in.  
 a = length of bearing surface one side of outer body, in.  
 g = gap between bearing surfaces, in.  
 L = total active length of pin, in.  
 =  $l + 2(a + g)$

P = 345,000 lb.  
 d = 7.495 in.  
 l = 8.00 in.  
 a = 3.00 in.  
 g = .1875 in.  
 $[14375 - 2(3.00) - 8] / 2$

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = 345,000$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (7.495)^2 / 4$$

$$A_v = 44.120 \text{ in}^2$$

$$f_v = (345,000) / (2 * 44.120)$$

$$f_v = \underline{\underline{3910 \text{ PSI}}}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = 345,000$$

INNER

$$A_v = d l = (7.495 * 8.00)$$

$$= (59.96) \text{ in}^2$$

$$f_c = (345,000) / (59.96)$$

$$f_c = \underline{\underline{5754 \text{ PSI}}}$$

OUTER

$$A_v = 2ad = 2(3.00)(7.495)$$

$$= (44.97) \text{ in}^2$$

$$f_c = (345,000) / (44.97)$$

$$f_c = \underline{\underline{7672 \text{ PSI}}}$$

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PIN

①

BENDING STRESS (2)

$$M = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right)$$

$$f_b = Mc/I$$

$$I = \pi d^4 / 64$$

$$c = d/2$$

$$f_b = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right) \left( \frac{d}{2} \right) \left( \frac{64}{\pi d^4} \right)$$

$$= 16P \left( \frac{1}{3} a + g + \frac{1}{4} l \right) / (\pi d^3)$$

$$= 16P \left( \frac{2.60}{3} + .1875 + \frac{7.00}{4} \right) / \pi 7.495^3$$

$$f_b = P (.03856)$$

$$= 345,000 (.03856)$$

$$= 13,303 \text{ psi}$$

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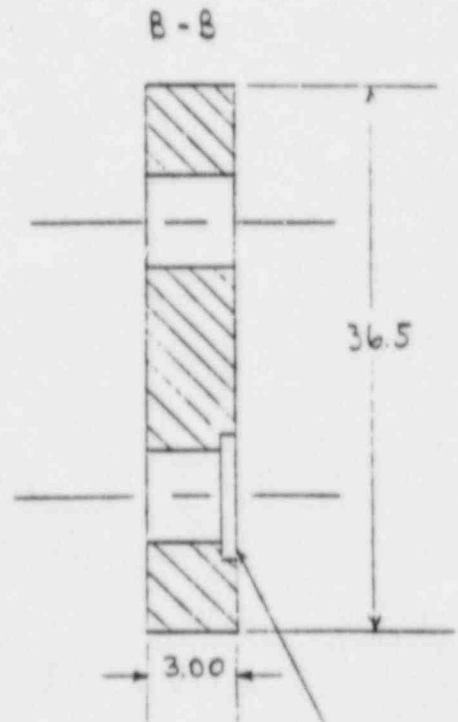
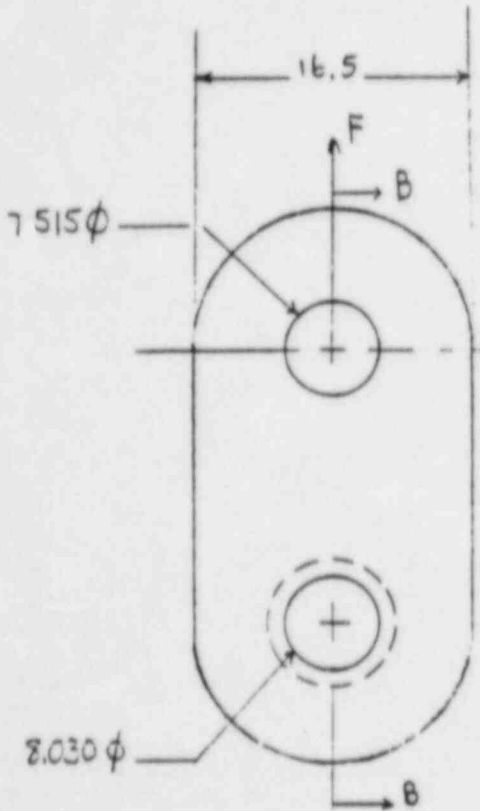


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SIDE PLATE ASTM A-514 or USS - T1

2



Tension at 8.030 Hole

$$F = \frac{345,000}{2} = 172,500$$

8.56  $\phi$  x .50 DEEP  
Counterbore on  
one plate

$$A = (16.5 - 8.030)(3.00) - (8.56 - 8.03)(.50) = 25.145 \text{ in}^2$$

$$f_t = \frac{F}{A} = \frac{172,500}{25.145}$$

$$f_t = 6860 \text{ psi}$$

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SO. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE		

Shear at 8.030 Hole

$$f_v = \frac{E}{2A_v} \quad A_v = \left( \frac{16.50 - 8.030}{2} \right) (13.00) - \left( \frac{7.56 - 8.030}{2} \right) (1.50)$$

$$f_v = \frac{172,500}{2(12.573)} \quad , \quad A_v = 12.573 \text{ in}^2$$

$$f_v = 6860 \text{ PSI}$$

Bearing at 7.515"  $\phi$  Hole

The maximum bearing stress is the same as the outer bearing stress on the hook pin ①.

$$f_c = (345,000) / 44.97 = 7672 \text{ PSI}$$

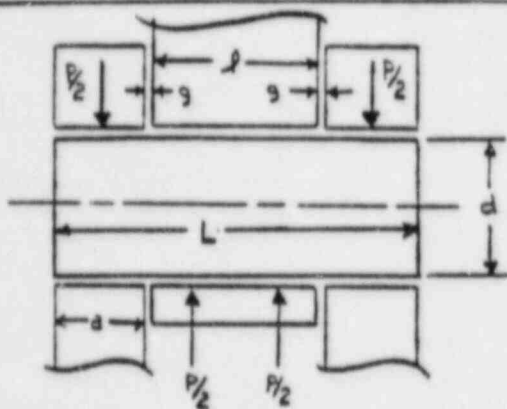
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8" DIA.  
PIN

3

MAT'L: ASTM A434 CLASS BD



P = force acting on assembly, lb.  
 d = diameter of pin, in.  
 l = length of bearing surface of center body, in.  
 a = length of bearing surface one side of outer body, in.  
 g = gap between bearing surfaces, in.  
 L = total active length of pin, in.  
 = l + 2(a + g)

P = 345,000 lb.  
 d = 8.00 in.  
 l = 8.06 in 7.15 - 2(0.45)  
 a = 3.00 in  
 g = .1575 in  
 [14.375 - 2(3.00) - 8.06] / 2

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = 345,000$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (8.000)^2 / 4$$

$$A_v = 50.27 \text{ in}^2$$

$$f_v = (345,000) / (2 * 50.27)$$

$$f_v = \underline{3431 \text{ PSI}}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = 345,000$$

INNER

$$A_v = d l = (8.00 * 8.06)$$

$$= (64.48) \text{ in}^2$$

$$f_c = (345,000) / (64.48)$$

$$f_c = \underline{5350 \text{ PSI}}$$

OUTER

$$A_v = 2ad = 2(3.00)(8.00)$$

$$= (48.0) \text{ in}^2$$

$$f_c = (345,000) / (48.0)$$

$$f_c = \underline{7187 \text{ PSI}}$$

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PIN

3

BENDING STRESS<sup>(2)</sup>

$$M_{max} = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right)$$

$$f_b = M c / I$$

$$I = \pi d^4 / 64$$

$$c = d / 2$$

$$f_b = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right) \left( \frac{d}{2} \right) \left( \frac{64}{\pi d^4} \right)$$

$$= 16P \left( \frac{1}{3} a + g + \frac{1}{4} l \right) / (\pi d^3)$$

$$= 16P \left( \frac{3.00}{3} + .1575 + \frac{8.06}{4} \right) / \pi 8.00^3$$

$$= P (.03156)$$

$$= 345,000 (.03156)$$

$$f_b = 10,878 \text{ PSI}$$

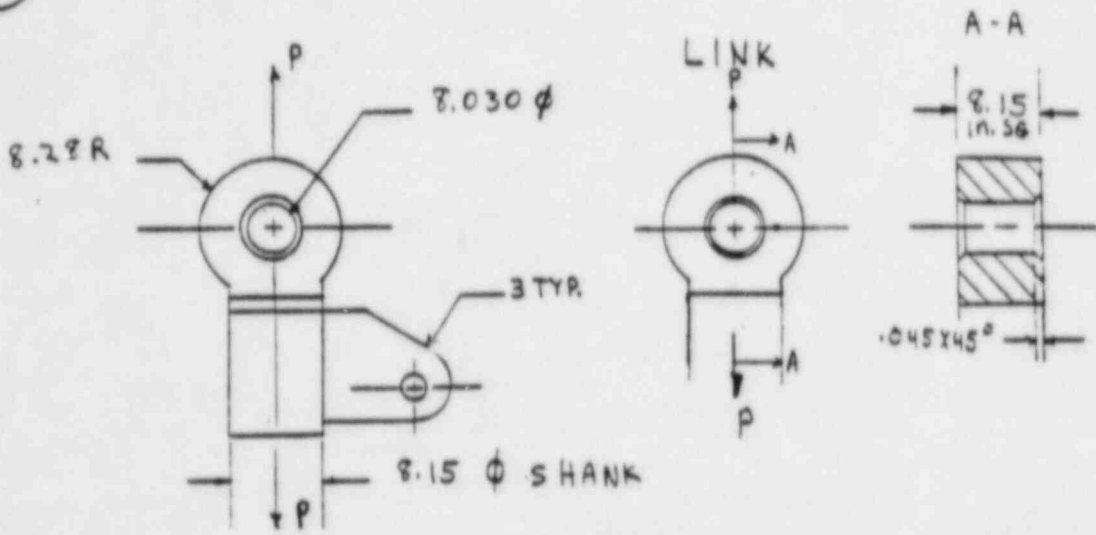
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SO PTNP-188	CALC. NO.	PL. NO.	GROUP CHE

4 SLING ASSEMBLY LINK ASTM A-237 CLASS A



TENSION at 8.030  $\phi$  Hole

$$f_t = \frac{P}{A_t} \quad A_t = (8.28 - 4.015)(8.15)(2) - \frac{1}{2} 4(.045)^2 = 69.515 \text{ in}^2$$

$$F_t = \frac{345,000}{69.515}$$

$$F_t = 4963 \text{ PSI}$$

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PSE/PNJ'		G.C. Peduzzi		1/83		g.w. Richard							
SO		CALC NO		DRAWING NO		GROUP						CHE	
PTNP-188													

Shear at 8.030  $\phi$  Hole

$$f_v = \frac{P}{A_v} \quad A_v = (8.28 - 4.015)(8.15)(2) - \frac{1}{2}4(.045)^2 = 69.515 \text{ in}^2$$

$$f_v = \frac{345,000}{69.515} = 4963 \text{ PSI}$$

Bearing on 8.00  $\phi$  Pin

The bearing stress is the same as the inner bearing stress on the bottom pin connected through the side plate. The pin is item (3).

$$f_c = 345,000 / 64.48 = 5350 \text{ PSI}$$

Tension at cylindrical section

$$f_t = \frac{P}{A_t} \quad A_t = \frac{\pi (8.15)^2}{4} = 52.168 \text{ in}^2$$

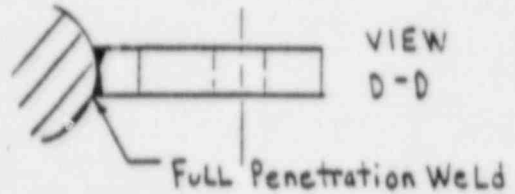
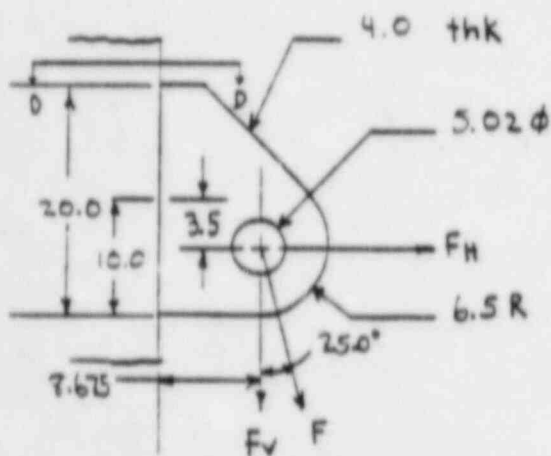
$$f_t = \frac{345,000}{52.168} = 6613 \text{ PSI}$$

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5

LINK LUG ASTMA237 CLASS A



$$F_v = \frac{345,000}{3} = 115,000 \text{ lbs}$$

$$F = \frac{F_v}{\cos 25^\circ} = 126,888 \text{ lbs}$$

$$F_H = F_v \tan 25^\circ = 53,625 \text{ lbs}$$

Tension at 5.02"  $\phi$  Hole

$$F_t = \frac{E}{A_t} \quad A_t = 2(6.5 - 2.51)4.0 = 31.92 \text{ in}^2$$

$$F_t = \frac{126,888}{31.92} = 3975 \text{ PSI}$$

Shear Tear at hole

$$F_v = \frac{E}{A_v} = \frac{126,888}{31.92} \quad A_v = 2(6.5 - 2.51)4 = 31.92 \text{ in}^2$$

$$F_v = 3975 \text{ PSI}$$

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SG PTNP-188	CALC NO	FILE NO	GROUP CHE
<p>Bearing at 5.02" <math>\phi</math> hole</p> <p>The bearing stress is the same as the inner bearing stress on the upper clevis pin (6).</p> <p><math>f_c = 126,888 / 20.01 = 6341 \text{ PSI}</math></p> <p>Combined Stress from Tension and Bending</p> <p><math>M = 8.675 F_v - 3.5 F_H</math></p> <p><math>M = (8.675)(115,000) - (3.5)(53,625)</math></p> <p><math>M = 809,938 \text{ in-lb.}</math></p> <p><math>f_t = \frac{E_H}{A} + \frac{M}{Z}</math></p> <p><math>A_f = (4)(20) = 80 \text{ in}^2</math></p> <p><math>Z = \frac{1}{6} (4)(20)^2 = 267 \text{ in}^3</math></p> <p><math>f_t = \frac{53625}{80} + \frac{809938}{267}</math></p> <p><math>f_t = 670 + 3033</math></p> <p><math>f_t = 3703 \text{ PSI}</math></p>			
REV. NO.	REV. DATE	AUTHOR	DATE
		CHK'D BY	DATE

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S.O.	CALC NO.		FILE NO.		GROUP
PTNP-188					CHE

vertical shear at FULL Penetration weld

$$f_v = \frac{E_v}{A_v} \quad A_v = (4)(20) = 80 \text{ in}^2$$

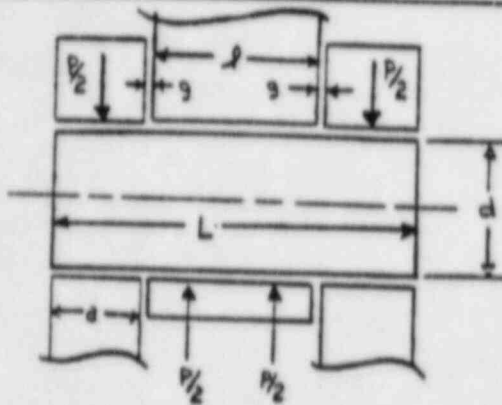
$$f_v = \frac{115,000}{80} = 1437 \text{ PSI}$$

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Upper Clevis 6  
PIN

MAT'L: ASTM A-434 CLASS BD



- P = force acting on assembly, lb.
- d = diameter of pin, in.
- l = length of bearing surface of center body, in.
- a = length of bearing surface one side of outer body, in.
- g = gap between bearing surfaces, in.
- L = total active length of pin, in.  
= l + 2(a + g)

P = 126,888 lb.  
 d = 5.0025 in.  
 l = 4.00 in.  
 a = 2.40 in 2.40 = 2(1.045)  
 g = .240 in  
 [9.37 - 2(2.40) + 2(1.045) - 4.00] / 2

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = 126,888$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (5.0025)^2 / 4$$

$$A_v = 19.65 \text{ in}^2$$

$$f_v = (126,888) / (2 \cdot 19.65)$$

$$f_v = \underline{3229 \text{ PSI}}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = 126,888$$

INNER

$$A_v = d l = (5.0025 \times 4.00)$$

$$= (20.01) \text{ in}^2$$

$$f_c = (126,888) / (20.01)$$

$$f_c = \underline{6341 \text{ PSI}}$$

OUTER

$$A_v = 2ad = 2(2.40)(5.0025)$$

$$= (24.012) \text{ in}^2$$

$$f_c = (126,888) / (24.012)$$

$$f_c = \underline{5284 \text{ PSI}}$$

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SO PTNP-188	CALC. NO.	LEN. NO.	GROUP CHE

PIN

(6)

BENDING STRESS (2)

$$M_{max} = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right)$$

$$f_b = M c / I$$

$$I = \pi d^4 / 64$$

$$c = d / 2$$

$$f_b = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right) \left( \frac{d}{2} \right) \left( \frac{64}{\pi d^4} \right)$$

$$= 16P \left( \frac{1}{3} a + g + \frac{1}{4} l \right) / (\pi d^3)$$

$$= 16P \left( \frac{2.40}{3} + .240 + \frac{4}{4} \right) / \pi 5.0003^3$$

$$= 126,888 (.08299)$$

$$= 10531 \text{ PSI}$$

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S.O. PTNP-188	CALL NO.	FILE NO.	GROUP CHE

⑦ UPPER & BOTTOM CLEVIS ASTM A-237 CLASS BD

⑨

Tension at 5.020  $\phi$  Hole

$$f_t = \frac{F}{A_t} = \frac{126,888}{2(19.946)}$$

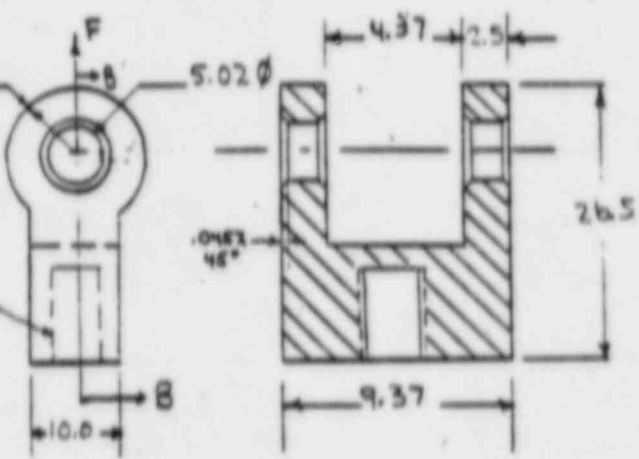
$$f_t = 3181 \text{ PSI}$$

Shear at 5.020  $\phi$  Hole

$$f_v = \frac{F}{A_v} = \frac{126,888}{4(9.973)}$$

$$f_v = 3181 \text{ PSI}$$

Bearing at 5.02"  $\phi$  Hole



$$A_t = (13 - 5.02)(2.5) - 4\left(\frac{1}{2} \times .045\right)^2$$

$$A_t = 19.946 \text{ in}^2$$

$$A_v = (6.5 - \frac{5.02}{2})(2.5) - 2\left(\frac{1}{2}\right)(.045)^2$$

$$A_v = 9.973 \text{ in}^2$$

The bearing stress is the same as the outer bearing stress on the clevis pin ⑥ for both ⑦ and ⑨.

$$f_c = 126,888 / 24.012 = 5284 \text{ PSI}$$

The thread shear is the same as the thread shear for the arm ⑧ for both ⑦ and ⑨.

$$f_v = 126,888 / 55.092 = 2303 \text{ PSI}$$

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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

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ARM

8

MAT'L: ASTM - A306 GR.70



THREAD SHEAR

$$f_v = P/A_v$$

$$A_v = D_{pitch} l / 2$$

FOR AN EXTERNAL THREAD

$$D_{pitch} = D_{nom} - .64952/n$$

$$D_{pitch} = (5) - .64952/14$$

$$= (4.8376) \text{ in.}$$

$D_{nom}$  = major diameter of external thread

$n$  = number of threads/in

$l$  = length of thread engagement = 7.25

THEFORE  $A_v = (55.092) \text{ in}^2$

$$P = 126,888 \text{ lbs}$$

$$f_v = 2303 \text{ PSI}$$

THREAD TENSION

$$f_t = P/A_t$$

FOR EXTERNALLY THREADED PARTS

$$A_t = \frac{\pi}{4} (D_{nom} - .9743/n)^2$$

$$= \frac{\pi}{4} (5 - .9743/14)^2 = 17.769 \text{ in}^2$$

$$f_t = (126,888) / (17.769)$$

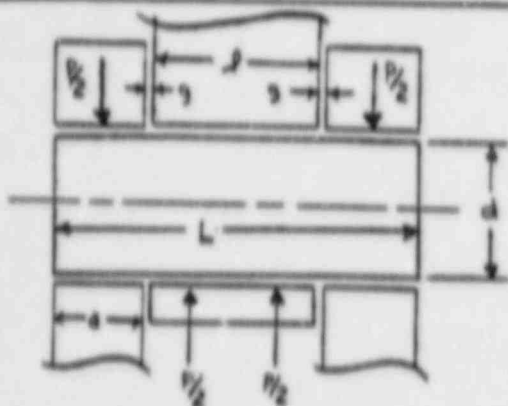
$$f_t = 7141 \text{ PSI}$$

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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

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Bottom Clevis (10)  
5" DIA. PIN  
MAT'Ls ASTM A434 CLASS 8D



P = force acting on assembly, lb.  
 d = diameter of pin, in.  
 l = length of bearing surface of center body, in.  
 a = length of bearing surface one side of outer body, in.  
 g = gap between bearing surfaces, in.  
 L = total active length of pin, in.  
 =  $l + 2(a + g)$

P = 126,888 lb.  
 d = 5.0025 in.  
 l = 3.91 in  $2(1.25) + 1.50 - 2(0.45)$   
 a = 2.40 in  $2.49 - 2(0.45)$   
 g = .285 in  
 $(9.37 - 2(0.48)) + 2(0.45 - 3.91)/2$

SHEAR STRESS

$$f_v = P/2A_v$$

$$P = 126,888$$

$$A_v = \pi d^2/4$$

$$A_v = \pi(5.0025)^2/4$$

$$A_v = 19.65 \text{ in}^2$$

$$f_v = (126,888)/(2 \times 19.65)$$

$$f_v = \underline{3229 \text{ PSI}}$$

BEARING STRESS

$$f_c = P/A_v$$

$$P = 126,888$$

INNER

$$A_v = d l = (5.0025 \times 3.91)$$

$$= (19.560) \text{ in}^2$$

$$f_c = (126,888)/(19.560)$$

$$f_c = \underline{6477 \text{ PSI}}$$

OUTER

$$A_v = 2ad = 2(2.40)(5.0025)$$

$$= (24.012) \text{ in}^2$$

$$f_c = (126,888)/(24.012)$$

$$f_c = \underline{5284 \text{ PSI}}$$

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PIN

(10)

BENDING STRESS (2)

$$M = \frac{F}{2} \left( \frac{3}{4} a + g + \frac{1}{4} l \right)$$

$$f_b = M c / I$$

$$I = \pi d^4 / 64$$

$$c = d / 2$$

$$f_b = \frac{F}{2} \left( \frac{3}{4} a + g + \frac{1}{4} l \right) \left( \frac{d}{2} \right) \left( \frac{64}{\pi d^4} \right)$$

$$= 16P \left( \frac{3}{4} a + g + \frac{1}{4} l \right) / (\pi d^3)$$

$$= 16P \left( \frac{2.40}{3} + 2.85 + \frac{2.91}{4} \right) / \pi (2.0625)^3$$

$$= 126,877 (.08391)$$

$$= 10,647 \text{ PSI}$$

(2) ADAPTED FROM  
FASTENING AND JOINING,  
4th Ed, A REFERENCE GUIDE OF  
MACHINE DESIGN, PUNCH PUBLISHING  
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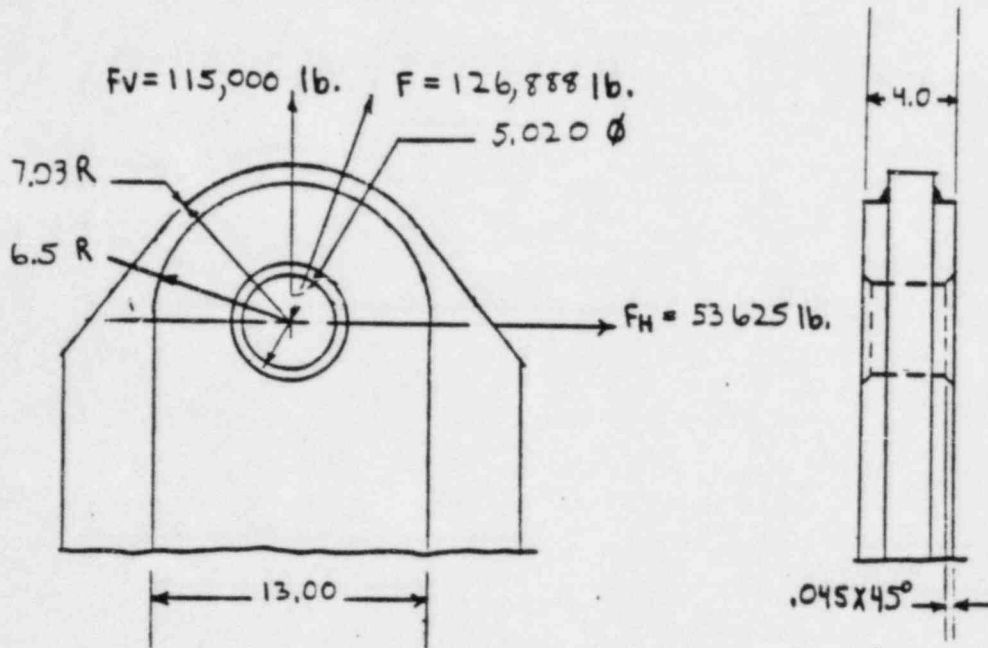
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11 SUPPORT LUG ASTM A-515 GR. 70 NET



Tension at Pin Hole

$$f_t = \frac{F}{A_t} \quad A_t = (13.00 - 5.02)4 - 4\left(\frac{1}{2}\right)(.045)^2 = 31.916 \text{ in}^2$$

$$f_t = \frac{126,888}{31.916} = 3976 \text{ PSI}$$

Shear at Pin Hole

$$f_s = \frac{F}{2A_v} = \frac{126,888}{2 \times 31.916} = 3976 \text{ PSI}, \quad A_v = (6.5 - 2.51)(4) - 2\left(\frac{1}{2}\right)(.045)^2$$

$$A_v = 15.957 \text{ in}^2$$

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S.O.	CALC. NO.	FILE NO.	GROUP			
PTNP-188			CHE			

Bearing at 5.02" hole

The bearing stress is the same as the inner bearing stress on the clevis pin (10).

$$f_c = 126,888 / 19.560 = 6487 \text{ PSI}$$

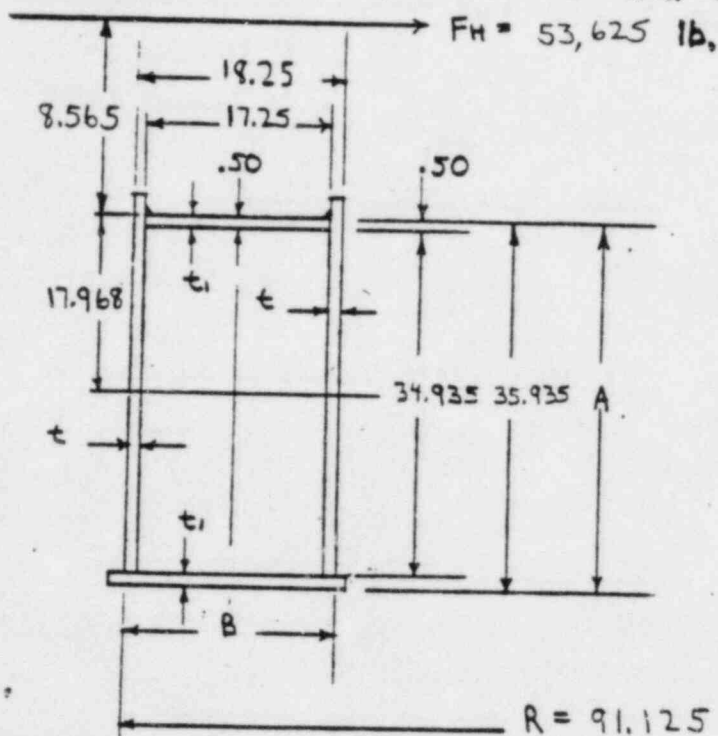
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12

RING GIRDER ASTM A-285 GR. C



TORSION

$$f_{TOR} = \frac{T}{2t(A-t)(B-t_1)} \quad \text{Roark 3RD EDITION Page 176, Case 11}$$

$$T = (8.565 + 17.968)(53,625)$$

$$T = 1,422,832 \text{ lb-in.}$$

$$f_{TOR} = \frac{1,422,832}{(2 \times .5) \times (35.435 - .5) \times (18.25 - .50)} = 2262 \text{ PSI}$$

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shear

$$f_v = \frac{F_H}{A} = \frac{53625}{(35.935)(18.25) - (34.935)(17.25)} = \frac{53625}{53.185}$$

$$f_v = 1008 \text{ PSI}$$

$$\text{Total Shear} = f_{TOR} + f_v = 2262 + 1008$$

$$\text{Total Shear} = 3270 \text{ PSI}$$

Bending stress Roark p.158, case 9

$$M_{max} = \frac{1}{2} WR \left( \frac{1}{s} - \frac{1}{\theta} \right) \text{ Between Loads } w$$

$$M_{max} = \frac{1}{2} (53625)(91.125) \left( \frac{1}{\sin \theta} - \frac{1}{\theta} \right)$$

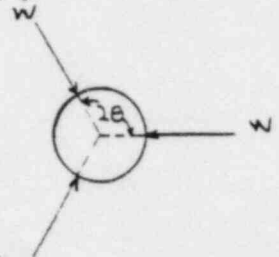
$$\theta = 60^\circ, \theta = 1.047 \text{ Rad. } \frac{1}{\sin \theta} = 1.154 \quad \frac{1}{\theta} = .955 \text{ Rad.}$$

$$M_{max} = (53625)(91.125)(1.154 - .955)$$

$$M_{max} = 972429 \text{ lb-in.}$$

$$-M_{max} = -\frac{1}{2} WR \left( \frac{1}{\theta} - \cot \theta \right), \text{ at each load}$$

$$\theta = 60^\circ, \theta = 1.047 \text{ Rad, } \frac{1}{\theta} = .955 \text{ Rad.}$$



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$$\cos \theta = .577$$

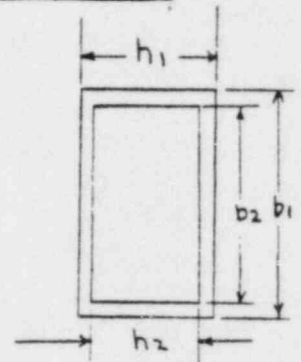
$$-M_{max} = -\frac{1}{2}(53625)(91.125)(.955 - .577)$$

$$-M_{max} = -923563 \text{ lb-in.}$$

$$Z = \frac{k_1 h^3 - k_2 h^3}{12(h/2)} = \frac{(35.935)(17.25)^3 - (34.935)(17.25)^3}{(6)(18.25)}$$

$$Z = 357 \text{ in}^3$$

$$f_B = \frac{M_{max}}{Z} = \frac{923563}{357} = 2587 \text{ PSI}$$



Maximum Tensile stress

$$f_{max} = \frac{f_B}{2} + \sqrt{\left(\frac{f_B}{2}\right)^2 + (f_v)^2}$$

$$f_{max} = \frac{2587}{2} + \sqrt{\left(\frac{2587}{2}\right)^2 + (3270)^2}$$

$$f_{max} = 1293.5 + \sqrt{1673142 + 10692900}$$

$$f_{max} = 1293.5 + 3517$$

$$f_{max} = 4811 \text{ PSI}$$

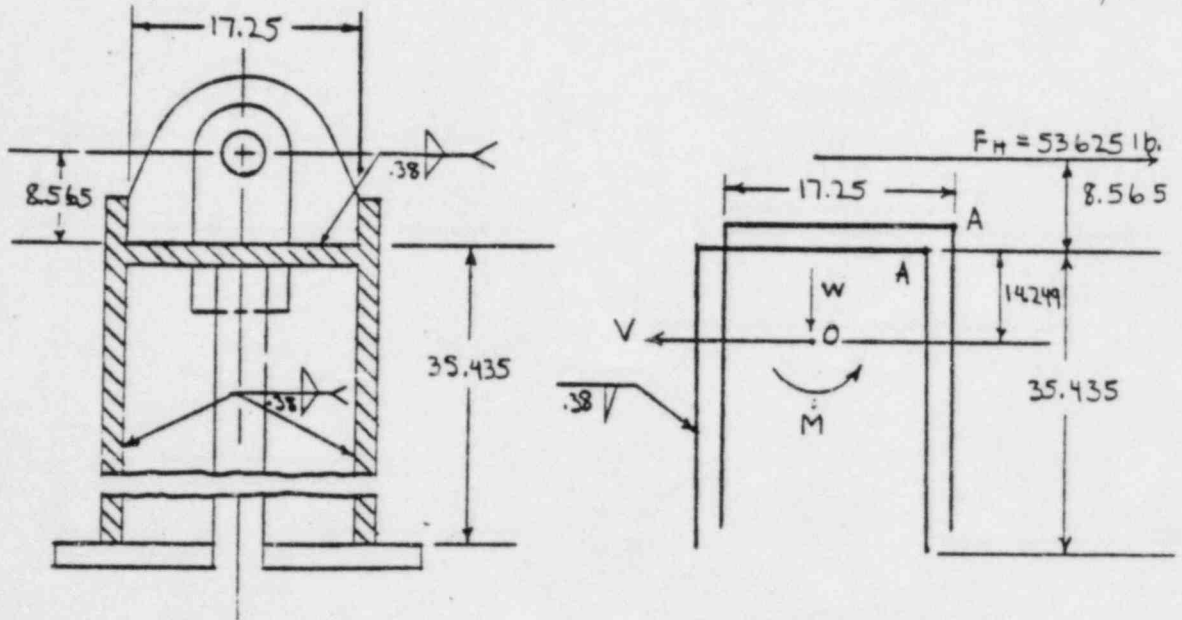
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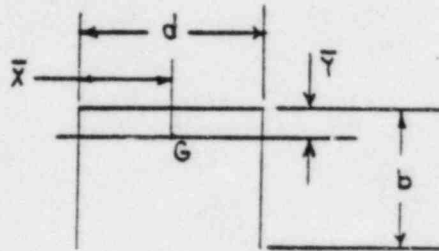
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WELD OF SUPPORT LUG TO RING GIRDER



Ref.: Shigley's Mechanical Engineering Design 3RD Edition



From Table 7-1

$$\bar{y} = \frac{b^2}{2b+d} = \frac{35.435^2}{2(35.435) + 17.25} = 14.249 \text{ in.}$$

$$\bar{x} = \frac{d}{2} = \frac{17.25}{2} = 8.625 \text{ in.}$$

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$$M = (F_H)(14.249 + 8.565)$$

$$M = (53625)(22.814) = 1223400 \text{ lb.-in.}$$

$$J = \frac{2b^3 + 6bd^2 + d^3}{12} - \frac{b^4}{2b+d}$$

$$b = 35.435", \quad d = 17.25"$$

$$J_U = \frac{2(35.435)^3 + 6(35.435)(17.25)^2 + 17.25^3}{12} - \frac{35.435^4}{2(35.435) + 17.25}$$

$$J_U = 35362 - 17892 = 17470 \text{ in}^3$$

$$J = [0.707 w J_U] 2$$

w = .38 fillet weld

$$J = 0.707 (.38)(17470) 2$$

$$J = 9387 \text{ in}^4$$

$$A_w = 0.707 w (2b + d)$$

$$A_w = (0.707)(.38)[2(35.435) + 17.25] 2 = 47.35 \text{ in}^2$$

$$\tau'_x = \frac{F_H}{A_w} = \frac{53625}{47.35} = 1133 \text{ PSI}$$

$$\tau'_y = \frac{W}{A_w} = \frac{5000}{47.35} = 106 \text{ PSI}$$

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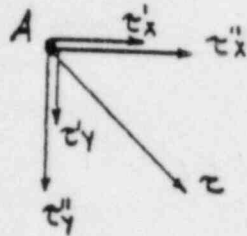
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w = 5000 (downward load)

$$\tau_y'' = \frac{Mx}{J} = \frac{(1223400)(8.625)}{9387} = 1124 \text{ PSI}$$

$$\tau_x'' = \frac{My}{J} = \frac{(1223400)(14.749)}{9387} = 1857 \text{ PSI}$$



Resultant stress in weld

$$\tau = \sqrt{\tau_y^2 + \tau_x^2}$$

$$\tau = \sqrt{(106 + 1124)^2 + (1133 + 1857)^2}$$

$$\tau = 3233 \text{ PSI}$$

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⑬ LEG ASTM A-36

Tension in Leg Tubing

$$F_t = \frac{F}{A_t}$$

For 8" x 6" x .50" wall Tubing,  $A_t = 11.9 \text{ in}^2$  (AISC)

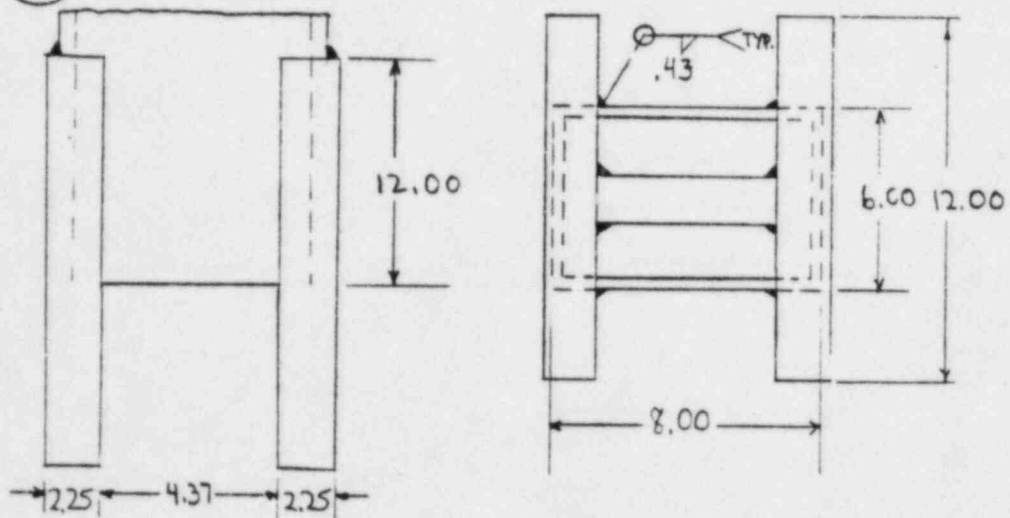
$$f_t = \frac{115,000}{11.9 \text{ in}^2} = 9664 \text{ PSI}$$

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14 CLEVIS PLATE A-515 GR. 70 Q + T



WELD OF CLEVIS PLATE TO LEG  
Length of Available Weld

$$l_a = 4(12) + 4(2.25 - .435) + 2(6) = 67.26$$

$$f_w = \frac{P}{l_a} = \frac{115,000}{67.26} = 1710 \text{ lb/in}$$

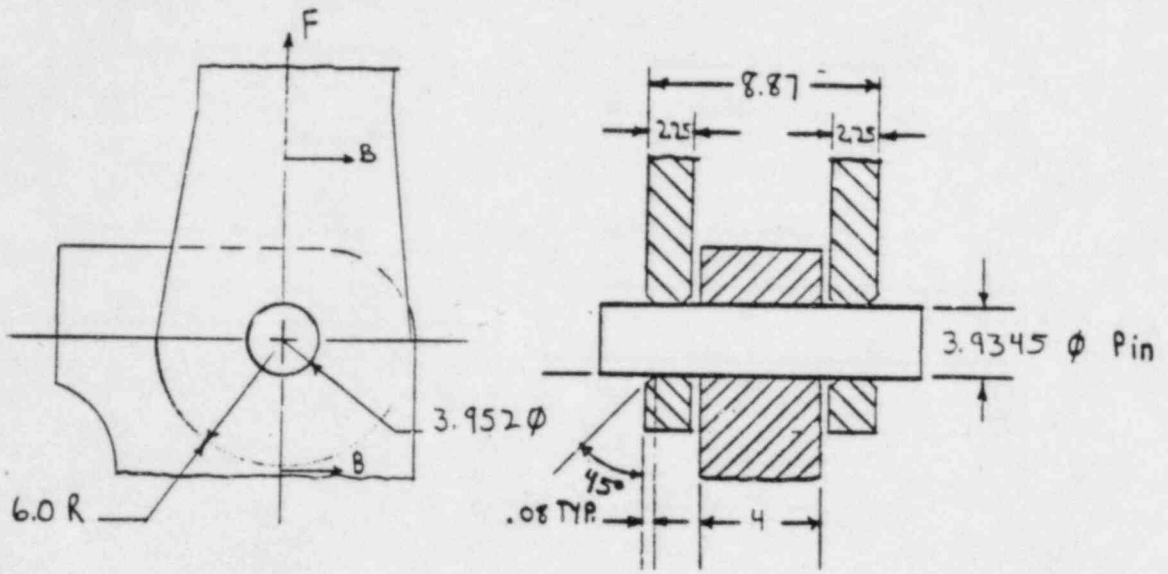
$$s_w = \frac{f_w}{(.707)(.43)} = \frac{1710}{(.707)(.43)} = 5625 \text{ PSI}$$

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Tension at Pin Hole

$$f_t = \frac{F}{A_t}, \quad A_t = ((12 - 3.952)(2.25) - 4(\frac{1}{2})(.08)^2)2 = 36.19 \text{ in}^2$$

$$f_t = \frac{115,000}{36.19} = 3178 \text{ PSI}$$

Shear at Pin Hole

$$f_v = \frac{F}{2A_v}, \quad A_v = ((6 - 1.976)(2.25) - 2(\frac{1}{2})(.08)^2)2 = 18.09 \text{ in}^2$$

$$f_v = \frac{115,000}{2(18.09)} = 3178 \text{ PSI}$$

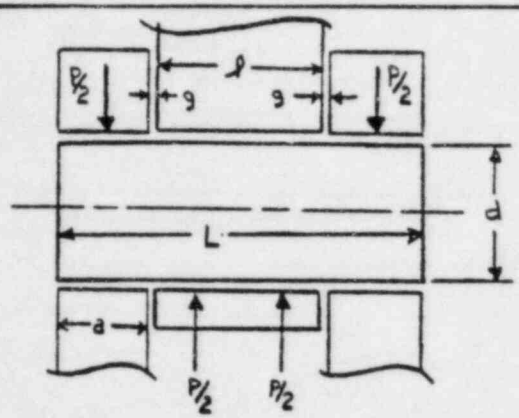
The Bearing stress is the same as the outer bearing stress on the 3.9345 PIN (15)  $f_c = 7307 \text{ PSI}$

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3.9345"  $\phi$  PIN 15  
 MAT'L: ASTM A434 CLASS BD



$P$  = force acting on assembly, lb.  
 $d$  = diameter of pin, in.  
 $l$  = length of bearing surface of center body, in.  
 $a$  = length of bearing surface one side of outer body, in.  
 $g$  = gap between bearing surfaces, in.  
 $L$  = total active length of pin, in.  
 $= l + 2(a + g)$

$P = 115,000$  lb.  
 $d = 3.9345$  in.  
 $l = 4.00$  nom. in.  
 $a = 2.09$  in  $2.25 - 2(.08)$   
 $g = .265$  in  
 $[4.37 + 2(.08) - 4] / 2$

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = 115,000$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (3.9345)^2 / 4$$

$$A_v = 12.158 \text{ in}^2$$

$$f_v = (115,000) / (2 * 12.158)$$

$$f_v = \underline{\underline{4729 \text{ PSI}}}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = 115,000$$

INNER

$$A_v = d l = (3.9345)(4.00)$$

$$= (15.738) \text{ in}^2$$

$$f_c = (115,000) / (15.738)$$

$$f_c = \underline{\underline{7307 \text{ PSI}}}$$

OUTER

$$A_v = 2ad = 2(2.09)(3.9345)$$

$$= (16.446) \text{ in}^2$$

$$f_c = (115,000) / (16.446)$$

$$f_c = \underline{\underline{6993 \text{ PSI}}}$$

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		A. C. Peduzzi	1/83	J. W. Richard	1/83				
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PIN 15

---

BENDING STRESS (2)

$$M_{max} = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right)$$

$$f_b = Mc / I$$

$$I = \pi d^4 / 64$$

$$c = d / 2$$

$$f_b = \frac{P}{2} \left( \frac{1}{3} a + g + \frac{1}{4} l \right) \left( \frac{d}{2} \right) \left( \frac{64}{\pi d^4} \right)$$

$$= 16P \left( \frac{1}{3} a + g + \frac{1}{4} l \right) / (\pi d^3)$$

$$f_b = 16P \left( \frac{2.09}{3} + .265 + \frac{4}{4} \right) / \pi 3.9345^3$$

$$f_b = 115,000 (.16403)$$

$$f_b = 18,863 \text{ PSI}$$
  
  

(1) ADAPTED FROM  
FASTENING AND JOINING,  
 4<sup>th</sup> Ed, A REFERENCE ISSUE OF  
 MACHINE DESIGN, PENTON PUBLISHERS  
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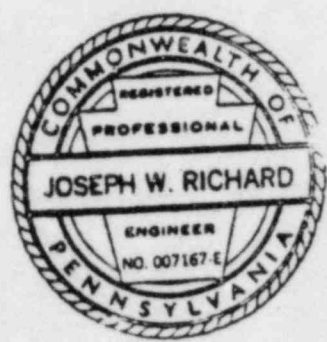
APPENDIX B  
DETAILED STRESS ANALYSIS - REACTOR VESSEL  
INTERNALS LIFT RIG, LOAD CELL AND LINKAGE

This appendix provides the detailed stress analysis for the Salem reactor vessel internals lift rig, load cell and linkage, in accordance with the requirements of ANSI N14.6. Acceptance criteria used in evaluating the calculated stresses are based on the material properties given in section 4.

SKETCH SHEET  
WESTINGHOUSE FORM 54202

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TITLE R.V. Internals Lift Rig		CALCULATIONS NO. PDC-
AUTHOR & DATE J. S. Urban <i>J.S. Urban</i>		CHECKED BY & DATE J. Richard <i>J. Richard 2/83</i>
PURPOSE AND RESULTS:		

1. The purpose of this analysis is to determine the acceptability of this rig to the requirements of ANSI N14.6.
2. The results show that all stresses are within the allowable stresses.



*J.W. Richard*

		Original Issue	J. Urban
REVISION NO.	DATE	DESCRIPTION	BY

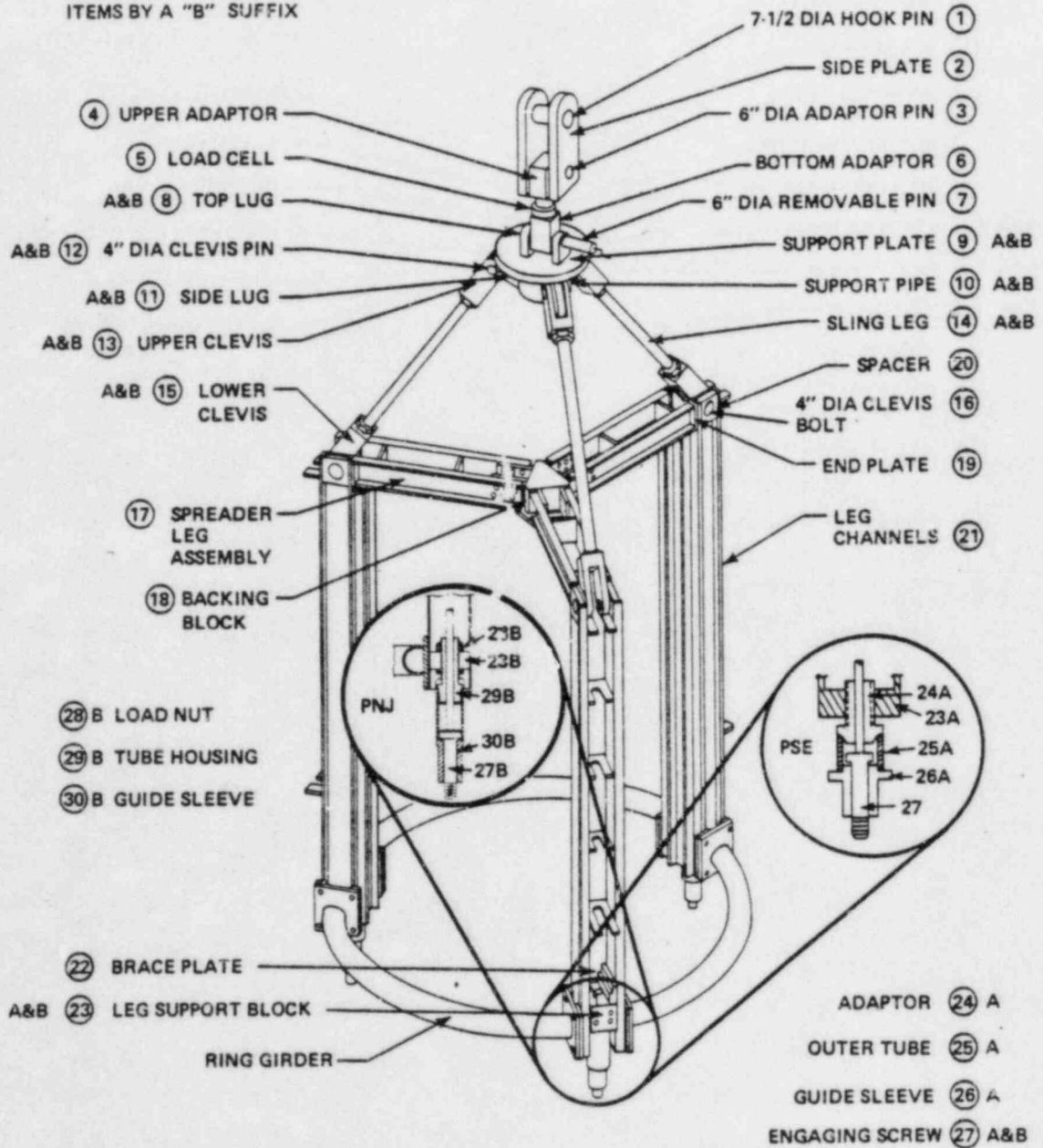
RESULTING REPORTS, LETTERS OR MEMORANDA:



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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP REE		

WHERE PSE AND PNJ RIGS DIFFER, THE PSE ITEMS ARE DISTINGUISHED BY AN "A" SUFFIX, THE PNJ ITEMS BY A "B" SUFFIX



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DESIGN WEIGHTS:

WEIGHT OF LOWER INTERNALS =	260,000 #
INTERNALS LIFTING RIG WEIGHT	14,500 #
CONTINGENCIES	10,000 #
	<u>284,500 #</u>

THE DESIGN WEIGHT FOR ITEMS  
1 THROUGH 6 WILL BE 320,000 lb  
AS THEY ARE ALSO USED TO MONITOR  
THE HEAD LIFT RIG LIFT

NOMENCLATURE:

$A$  = area, in<sup>2</sup>  
 $C$  = distance from center of flexure to extreme fibers, in  
 $d$  = diameter, in.  
 $f_b$  = bending stress, psi  
 $f_c$  = bearing stress, psi  
 $f_t$  = tensile stress, psi  
 $f_y$  = shear stress, psi  
 $I$  = moment of inertia, in<sup>4</sup>  
 $K$  = outward force exerted by spreader, = 71,752 #  
 $L$  = tension in lifting leg = 94,833 #  
 $M$  = moment, in-lb  
 $P$  = general force, lb  
 $T$  = tension in sling leg, = 118,916 #  
 $W$  = design weight  
 $\alpha$  = angle sling leg makes to vertical

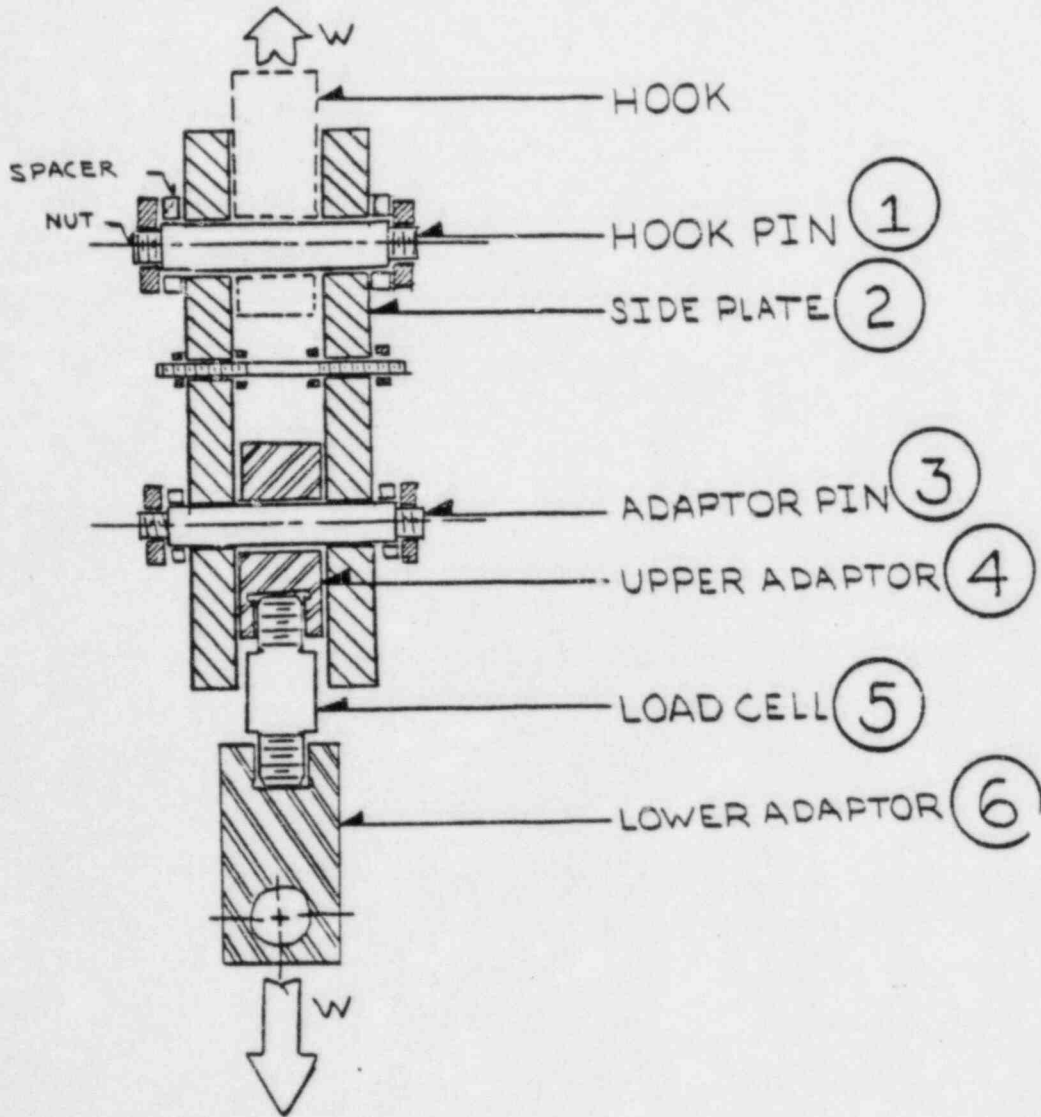
The nomenclature of references or variables defined at a calculation may be used as supplements to the above list. Where this is so, the meanings of the terms are defined where the calculations are made.

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LOAD CELL LINKAGE ASSEMBLY

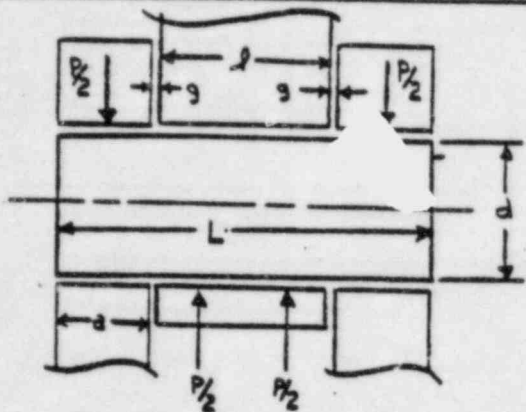


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**HOOK PIN** (1)

MAT'L: ASTM A 434 CLASS BD  
AISI 4340 STEEL



$P$  = force acting on assembly, lb.  
 $d$  = diameter of pin, in.  
 $l$  = length of bearing surface of center body, in.  
 $a$  = length of bearing surface one side of outer body, in.  
 $g$  = gap between bearing surfaces, in.  
 $L$  = total active length of pin, in.  
 $= l + 2(a + g)$

$P = W$  lb. = 320,000  
 $d = 7.500 / 7.492$  in.  
 $l = 8.23$  in  
 $a = 2.04 / 1.96$  in  
 $g = 0.08$  in  
 $L = 13.77 / 13.72$  in

**SHEAR STRESS**

$$f_v = P / 2A_v$$

$$P = W$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (7.492)^2 / 4$$

$$A_v = 44.084 \text{ in}^2$$

$$f_v = (W) / (2 \cdot 44.084)$$

$$= W(0.011342)$$

$$f_v = \underline{3629 \text{ PSI}}$$

**BEARING STRESS**

$$f_c = P / A_v$$

$$P = W$$

**INNER**

$$A_v = d l = (7.492)(8.23)$$

$$= (61.66) \text{ in}^2$$

$$f_c = (W) / (61.66)$$

$$= W(0.016218)$$

$$f_c = \underline{5190 \text{ PSI}}$$

**OUTER**

$$A_v = 2ad = 2(1.96)(7.492)$$

$$= (29.369) \text{ in}^2$$

$$f_c = (W) / (29.369)$$

$$= W(0.03405)$$

$$f_c = \underline{10,896 \text{ PSI}}$$

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BENDING STRESS\*

$$f_b = (P/2) \left( \frac{a}{2} + g + \frac{d}{4} \right) \frac{32}{\pi d^3}$$

$$f_b = (W) / 2 \cdot (1.96 / 2 + 0.08 + 8.23 / 4) \cdot 32 / (\pi (7.492)^3)$$

$$= W (0.03776)$$

$$f_b = \underline{\underline{12,082 \text{ PSI.}}}$$

MISCELLANEOUS CALCULATIONS

$$2g = L - d - 2a - 2 \text{ spacers thickness}$$

$$2g = 13.77 - (8.23) - 2(1.96) - 2(.73)$$

$$g = 0.08$$

\* EQUATION DERIVED ON P(17)

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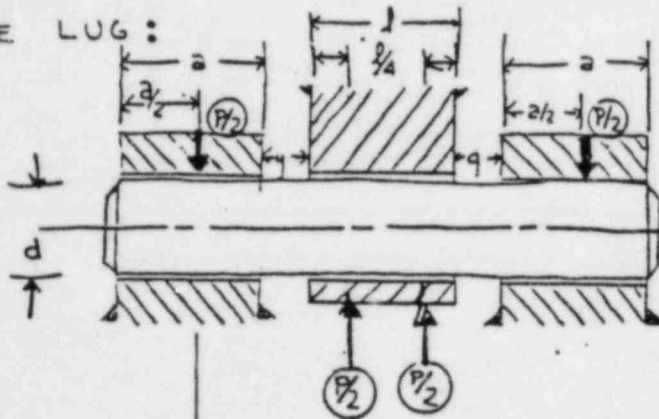


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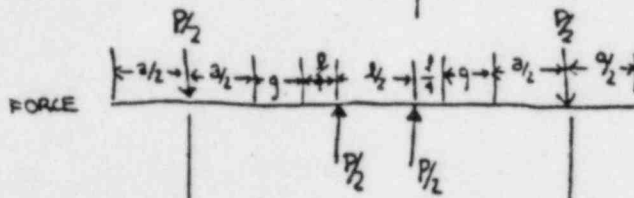
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— BENDING STRESS FORMULA DERIVATION —

ASSUMING FORCES IN DOUBLE LUG TO ACT AT THE LUG CENTERS AND THE FORCE IN THE CENTER LUG TO ACT AT TWO PLACES 1/4 WAY INTO THE LUG:



- $g$  = gap between bearing surfaces
- $a$  = length of one side of double-lug bearing surface
- $l$  = length of bearing surface of center lug
- $P$  = force acting on assembly
- $d$  = diameter of pin



$$f_b = \frac{Mc}{I}$$

$$c = \frac{d}{2}$$

$$I = \frac{\pi d^4}{64}$$

SHEAR



$$f_b = M_{max} [c / I]$$

$$f_b = \left(\frac{P}{2}\right) \left(\frac{a}{2} + g + \frac{l}{4}\right) \frac{32}{\pi d^3}$$

MOMENT



N.B. — This same maximum moment also occurs when the forces are assumed evenly distributed across the lug surfaces

$$M_{max} = \frac{P}{2} \left(\frac{a}{2} + g + \frac{l}{4}\right)$$

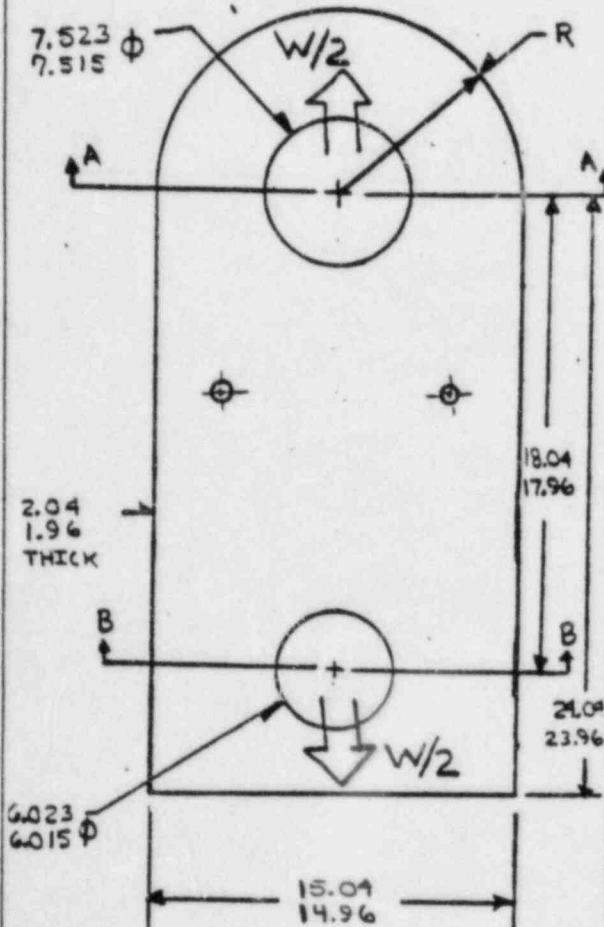
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SIDE PLATE (2)

MAT'L: ASTM-A515 GR. 70  
QUENCHED AND TEMPERED.



$W = 320,000 \#$

TENSION @ SEC A-A

$$f_t = P/A_t$$

$$P = W/2$$

$$A_t = (14.96 - 7.523) 1.96$$

$$= 14.577 \text{ in}^2$$

$$f_t = W(0.0343)$$

$$f_t = \underline{\underline{10,976 \text{ PSI}}}$$

BEARING @ SEC B-B

THE BEARING STRESS @ SEC B-B IS THE SAME AS THE OUTER BEARING STRESS ON THE ADAPTER PIN

$$f_c = W(0.04257)$$

$$f_c = \underline{\underline{13,622 \text{ PSI}}}$$

SHEAR TEAR-OUT @ B-B



CONSERVATIVELY USING THE CROSS-SECTION BELOW THE ANTS OF THE HOLE:

$$f_v = P/A_v$$

$$P = W/2$$

$$A_v = (23.96 - 18.04 - 6.023/2) 1.96$$

$$f_v = W(0.04385)$$

$$f_v = \underline{\underline{14,033 \text{ PSI}}}$$

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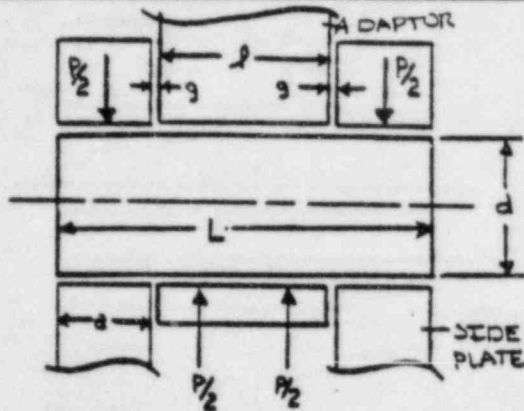
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ADAPTOR PIN

3

MAT'L: ASTM A434 CLASS BD  
AISI 4340 STEEL



P = force acting on assembly, lb.  
 d = diameter of pin, in.  
 l = length of bearing surface of center body, in.  
 a = length of bearing surface one side of outer body, in.  
 g = gap between bearing surfaces, in.  
 L = total active length of pin, in.  
 =  $l + 2(a + g)$

P = W lb. = 320,000 #  
 d = 6.000 // 5.992 in.  
 l = 8.27 // 8.23 in.  
 a = 2.00 ± .04 in.  
 g = 0.08 in.  
 L = 13.77 // 13.73 in.

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = W$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (5.992)^2 / 4$$

$$A_v = 28.199 \text{ in}^2$$

$$f_v = (W) / (2 * 28.199)$$

$$f_v = W(0.01773)$$

$$f_v = \underline{5674 \text{ PSI}}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = W$$

INNER

$$A_v = d l = (5.992 * 8.23)$$

$$= (49.314) \text{ in}^2$$

$$f_c = (W) / (49.314)$$

$$= W(0.02028)$$

$$f_c = \underline{6490 \text{ PSI}}$$

OUTER

$$A_v = 2ad = 2(1.96)(5.992)$$

$$= (23.49) \text{ in}^2$$

$$f_c = (W) / (23.49)$$

$$= W(0.04257)$$

$$f_c = \underline{13622 \text{ PSI}}$$

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BENDING STRESS\*

$$f_b = (P/2) \left( \frac{a}{2} + g + \frac{d}{4} \right) \frac{32}{\pi d^3}$$

$$f_b = (W) / 2 \cdot (1.96 / 2 + 0.08 + 8.23 / 4) = 32 / (\pi (5.992)^3)$$

$$= W (0.07380)$$

$$f_b = \underline{\underline{23,616 \text{ PSI.}}}$$

MISCELLANEOUS

$$2g = L - d - 2(a) - 2 \text{ spacer thicknesses}$$

$$2g = 13.77 - 8.23 - 2(1.96) - 2(.73)$$

$$g = 0.08$$

\* EQUATION DERIVED ON P(7)

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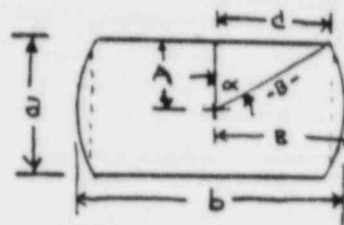
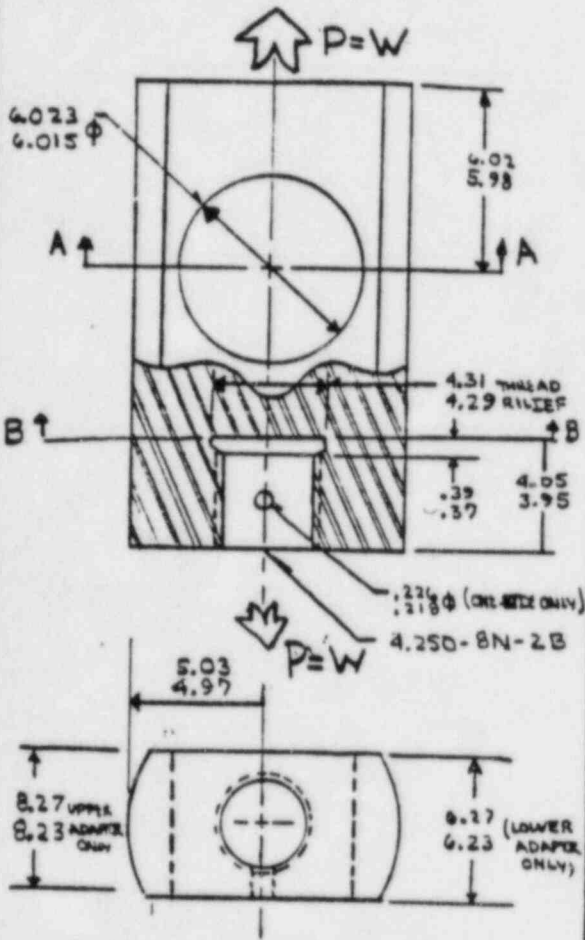
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LOWER ADAPTOR AND UPPER ADAPTOR

4/6

DWG 108D058 H01 AND H02

MAT'L: AISE 4340  
QUINCHED AND TEMPERED  
YIELD: 120,000 PSI  
TENSILE STRENGTH: 135,000 PSI.



BODY AREA  
 $A_B = 2AC + \pi(B^2)/4 + \frac{360-4\alpha}{360}$   
 $\cos \alpha = A/B$   
 $C = \sqrt{B^2 - A^2}$   
 $A = \frac{1}{2}a \quad B = \frac{1}{2}b$

---

$A_B = \frac{a}{2} \sqrt{b^2 - a^2} + \pi b^2 \left( \frac{360 - 4\alpha}{4(360)} \right)$   
 $\cos \alpha = a/b$

---

FOR LOWER ADAPTOR  
 $a = 6.23$   
 $b = 9.94$   
 $\therefore \alpha = 51.19^\circ$   
 $A_B = 57.59 \text{ in}^2$

FOR UPPER ADAPTOR  
 $a = 8.23$   
 $b = 9.94$   
 $\alpha = 34.11^\circ$   
 $A_B = 71.13 \text{ in}^2$

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LOWER ADAPTOR AND UPPER ADAPTOR  $\left(\frac{4}{6}\right)$

$$W = 320,000 \#$$

TENSILE STRESS @ A-A

$$A_t = A_B - A_{hole}$$

UPPER:  $A_t = 71.13 - 6.023 (8.23)$   
 LOWER:  $A_t = 57.59 - 6.023 (6.23)$   
 $f_t = P/A_t$   
 $P = W$

$$f_{tUPPER} = W (0.04638)$$

$$\underline{f_{tUPPER} = 14,842 \text{ PSI}}$$

$$f_{tLOWER} = W (0.04983)$$

$$\underline{f_{tLOWER} = 15,946 \text{ PSI}}$$

BEARING STRESS @ A-A

$$f_c = P/A_c$$

$$P = W$$

$$A_{cUPPER} = 6.023 (8.23)$$

$$= 49.569 \text{ in}^2$$

$$f_{cUPPER} = W (0.02017)$$

$$\underline{f_{cUPPER} = 6454 \text{ PSI}}$$

$$A_{cLOWER} = 6.023 (6.23)$$

$$= 37.52 \text{ in}^2$$

$$f_{cLOWER} = W (0.02665)$$

$$\underline{f_{cLOWER} = 8528 \text{ PSI}}$$

TENSILE STRESS @ B-B

$$f_t = P/A_t$$

$$A_t = A_B - \pi d^2/4$$

$$d_{RELIEF} = 4.31$$

$$A_{tUPPER} = 71.13 - \pi (4.31)^2/4 = 56.54$$

$$A_{tLOWER} = 57.59 - \pi (4.31)^2/4 = 43.00$$

$$P = W$$

$$f_{tUPPER} = W (0.017687)$$

$$\underline{f_{tUPPER} = 5660 \text{ PSI}}$$

$$f_{tLOWER} = W (0.02326)$$

$$\underline{f_{tLOWER} = 7443 \text{ PSI}}$$

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LOWER ADAPTOR  
AND UPPER ADAPTOR

4/6

THREAD SHEAR

$$f_v = P/A_v$$

$$A_v = \pi D_{\text{PITCH}} \cdot l/2$$

FROM MARKS HANDBOOK FOR M.E.'S  
8th ED, SEC 8, FOR A 1/4"-8N-2B

$$D_{\text{PITCH}} = 4.1688$$

$$l = 3.95 - .39 = 3.56 \text{ in}$$

$$A_v = \pi (4.1688) 3.56/2$$

$$= 23.31 \text{ in}^2$$

$$P = W$$

$$f_v = W(0.04290)$$

$$\underline{f_v = 13728 \text{ PSI}}$$

SHEAR TEAR-OUT



$$f_v = P/2A_v$$

CONSERVATIVELY, LET  $A_v$  BE THE AREA OF THE CROSS-SECTION DIRECTLY ABOVE THE AXIS.

FOR THE LOWER ADAPTOR

$$A_v = (5.98 - 6.023/2) \times (6.23)$$

$$= 18.494 \text{ in}^2$$

FOR THE UPPER ADAPTOR THE WIDTH OF THE MACHINED SURFACE IS  $2(4.97^2 - (8.23/2)^2)^{1/2}$

$$= 5.574 \text{ in. THIS IS LESS THAN THE HOLE } \phi = 6.023,$$

SO LET THE THICKNESS BE  $2(4.97^2 - (6.023/2)^2)^{1/2} = 7.907 \text{ in}$

INSTEAD OF 8.23 in.

$$A_v = (7.907 \times 5.98 - 6.023/2)$$

$$= 23.47 \text{ in}^2$$

$$P = W$$

$$f_{v \text{ UPPER}} = W/(2 \times 23.47)$$

$$f_{v \text{ UPPER}} = W(0.02130)$$

$$\underline{f_{v \text{ UPPER}} = 6816 \text{ PSI}}$$

$$f_{v \text{ LOWER}} = W/(2 \times 18.494)$$

$$f_{v \text{ LOWER}} = W(0.02704)$$

$$\underline{f_{v \text{ LOWER}} = 8653 \text{ PSI}}$$

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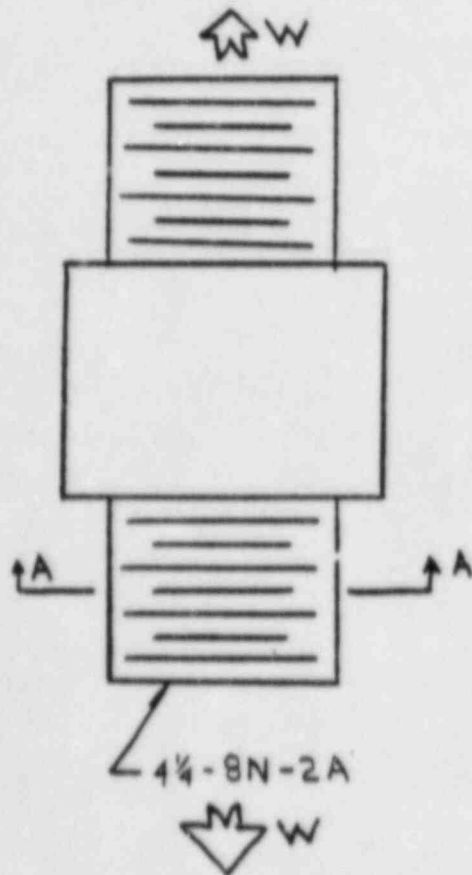
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LOAD CELL

5

MAT'L: 17-4 ph s/s  
COND H 1100°



W = 329,000 #

TENSION C A-A

$$f_t = P / A_t$$

$$P = W$$

FROM MARK'S HANDBOOK FOR M.E.S  
8th Ed, P8-13

STRESS AREA = 13.3683 =  $A_t$

$$f_t = W / 13.3683$$

$$f_t = W (0.07480)$$

$$f_t = \underline{\underline{23,936 \text{ PSI}}}$$

THREAD SHEAR

THE THREAD SHEARING STRESS  
IS THE SAME AS FOR THE  
ADAPTORS

$$f_v = W (0.04290)$$

$$f_v = \underline{\underline{13,728 \text{ PSI}}}$$

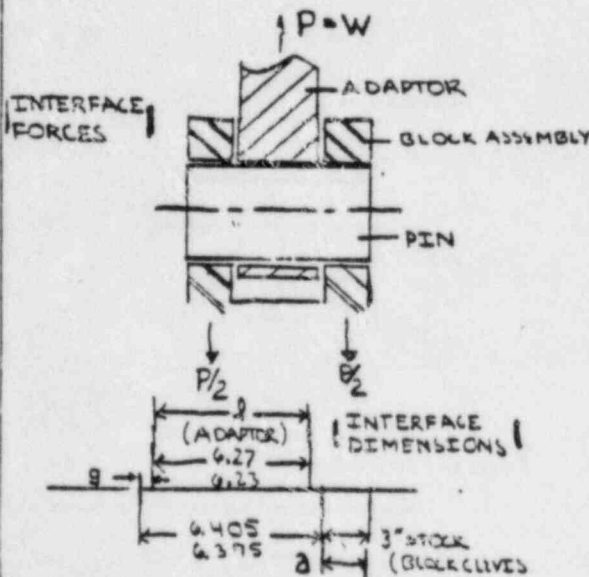
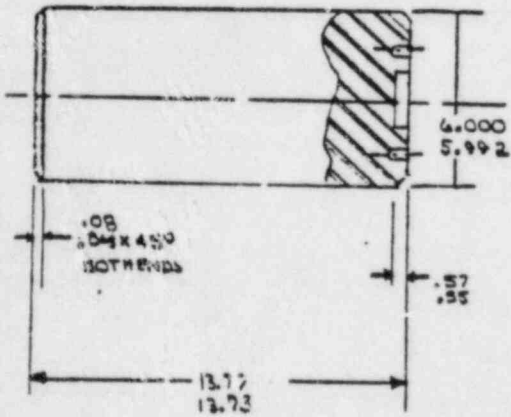
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REMOVABLE PIN (7)

EST WT: 111.0#  
MAT'L ASTM A434 CLASS BD  
AISI 4340



W = 284,500#  
SHEAR STRESS

$$f_v = (P/2) / A_v$$

$$A_v = \pi d^2 / 4$$

$$= \pi (5.992)^2 / 4$$

$$= 28.199 \text{ in}^2$$

$$(P/2) = W/2$$

$$f_v = W / (2 * 28.199)$$

$$= W (.017731)$$

$$f_v = \underline{\underline{5044 \text{ PSI}}}$$

BEARING STRESS

PIN TO BLOCK ASSEMBLY

$$f_c = (P/2) / A_c$$

$$A_c = dt = 3(5.992)$$

$$= 17.98 \text{ in}^2$$

$$P = W$$

$$f_c = W / (2 * 17.98)$$

$$f_c = W (.027815)$$

$$f_c = \underline{\underline{7913 \text{ PSI}}}$$

PIN TO ADAPTOR

$$f_c = P / A_c$$

$$P = W$$

$$A_c = dt = 5.992(6.23) = 37.33 \text{ in}^2$$

$$f_c = W (.026788)$$

$$\therefore f_c = \underline{\underline{7621 \text{ PSI}}}$$

REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig				PAGE 16 of 67	
PROJECT PSE/PNJ	AUTHOR <i>M. J. Law</i>	DATE 12/82	CHK'D. BY <i>J. W. Richard</i>	DATE 2/83	CHK'D. BY DATE
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE		

REMOVABLE PIN (7)

BENDING STRESS\*

$$P = W$$

$$a = 3. \text{ in}$$

$$l = 6.23 \text{ in}$$

$$g = (6.405 - 6.23) / 2 = .0875 \text{ in}$$

$$d = 5.992 \text{ in}$$

$$f_b = \left(\frac{P}{2}\right) \left(\frac{a}{2} + g + \frac{l}{4}\right) \frac{32}{\pi d^3}$$

$$= W \left(\frac{1}{2}\right) \left(3/2 + .0875 + 6.23/4\right)$$

$$\cdot 32 / \pi / d^3$$

$$f_b = W (.07445) \text{ PSI}$$

$$\underline{f_b = 21,181 \text{ PSI}}$$

THE FORMULA FOR PIN BENDING STRESS, USED ON THIS AND ALL SUBSEQUENT PINS, IS DERIVED ON THE FOLLOWING PAGE:

- PAGE 17 -

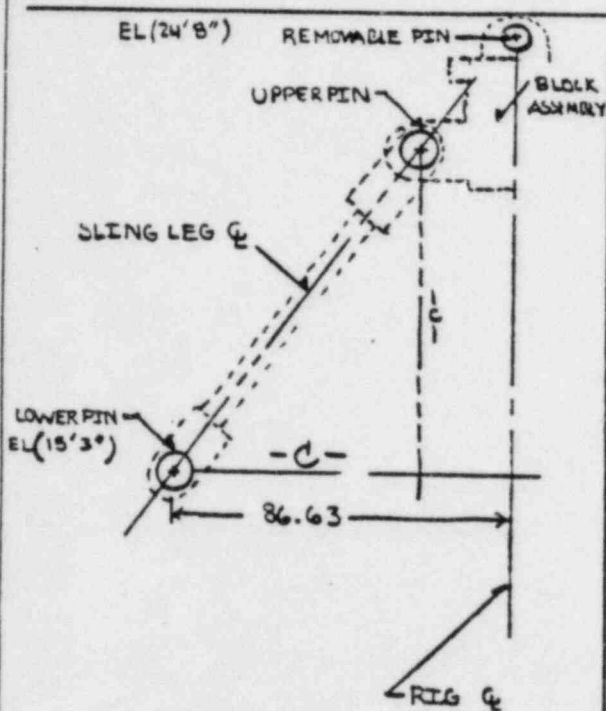
REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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PROJECT PSE/PNJ	AUTHOR <i>J. W. Richard</i>	DATE 12/82	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

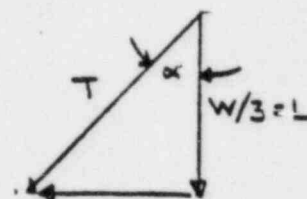
CALCULATION OF TENSION IN SLING LEG



$A = \text{RIG } Q \text{ TO LOWER PIN } Q = 86.63$   
 $B = \text{RIG } Q \text{ TO UPPER PIN } Q = 14.00$   
 $c = (A - B) = 72.63$   
 $b = \text{DIFFERENCE IN ELEVATION ABOVE HEAD OF REMOVABLE PIN AND LOWER PIN CENTERLINES}$   
 $= 24'8'' - 15'3'' = 9'5'' = 113''$   
 $b = \text{DIFFERENCE IN ELEVATION OF REMOVABLE PIN AND UPPER PIN (FROM BLOCK ASSY)}$   
 $= 7.0 + 19.50 - 1.0 - 9.5 = 17.0''$

$$c = a - b = 96''$$

$$\sqrt{72.63^2 + 96^2} = 120.38 \text{ (CALC)}$$



$$W = 284,500 \text{ lb}$$

W = WEIGHT OF LOWER INTERNALS + RIG WT.

T = TENSION IN SLING LEG

K = OUTWARD FORCE EXERTED BY SPREADER

L = DOWNWARD FORCE EXERTED BY A LEG = W/3

$$\tan \alpha = \frac{c}{b} = \frac{72.63}{96}$$

$$\alpha = 37.11^\circ$$

$$T \cos \alpha = L = W/3$$

$$T = (.4180)W$$

$$T = \underline{118,916 \text{ lb}}$$

$$T \sin \alpha = K$$

$$= .418 W \sin \alpha$$

$$\therefore K = (.252)W$$

$$K = \underline{71,751 \text{ lb}}$$

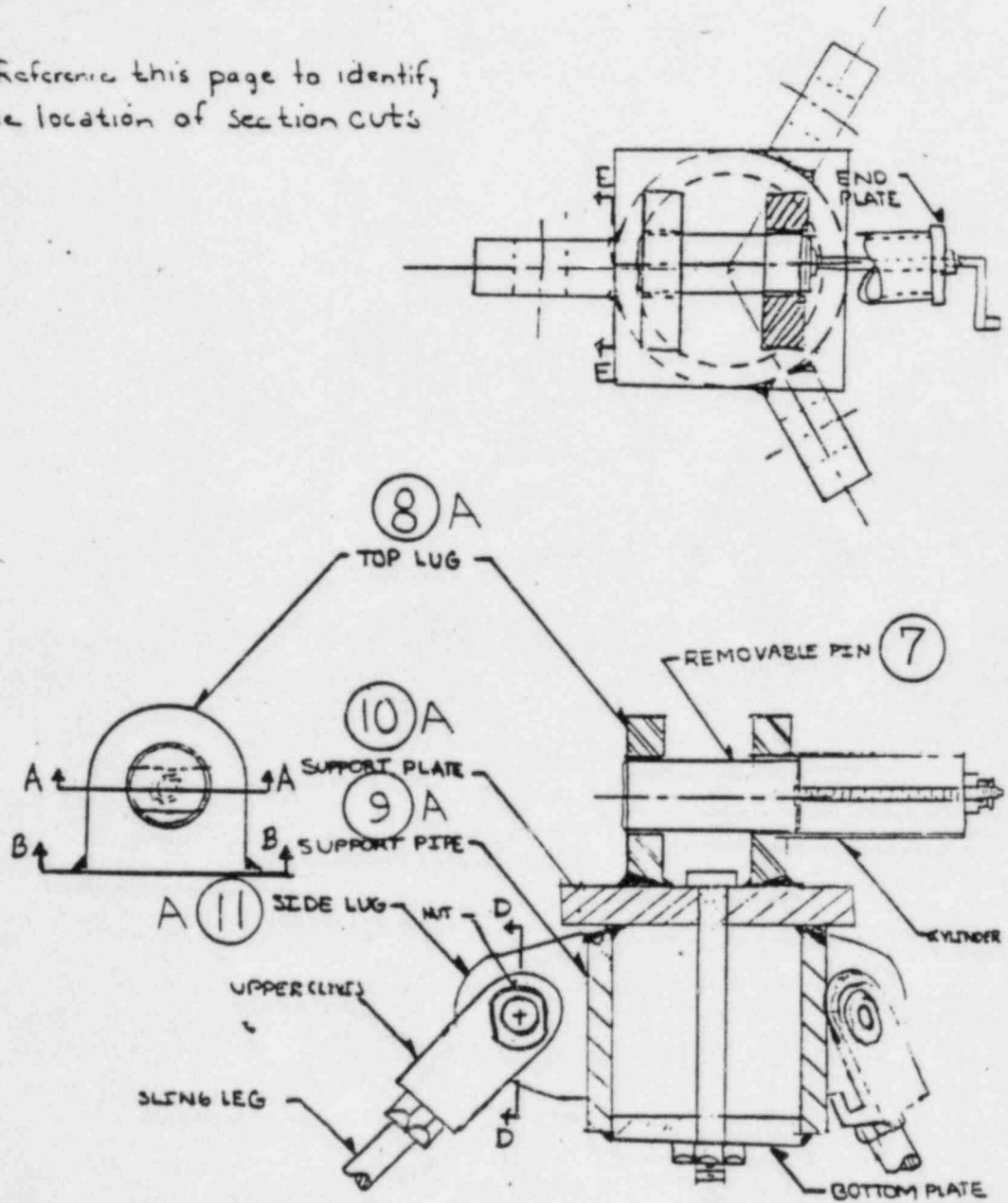
REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 18 of 67	
PROJEC: PSE/PNJ	AUTHOR <i>J. S. ...</i>	DATE 1/8/62	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CAUC. NO.	FILE NO.	GROUP CHE

• BLOCK ASSEMBLY • FSE

Reference this page to identify the location of section cuts



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TITLE R. V. Internals Lift Rig		PAGE 19 of 67	
PROJECT PSE/PNJ	AUTHOR <i>J. W. Richard</i>	DATE 2/8/67	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

### TOP LUG ⑧A

MAT'L  
ASTM A 515  
GR 70  
QUENCHED & TEMPERED

W = 284,500#

TENSION @ A-A

$$f_t = (P/2) / A_t$$

$$P = W$$

$$A_t = 3(11.94 - 6.028) = 17.736 \text{ in}^2$$

$$f_t = W (0.02819)$$

$$f_t = \underline{\underline{8020 \text{ PSI}}}$$
  

### BEARING @ A-A

THE BEARING STRESS IS THE SAME AS THE OUTER BEARING STRESS OF THE REMOVABLE PIN. (7)

$$f_c = W (.027815)$$

$$f_c = \underline{\underline{7913 \text{ PSI}}}$$
  

### TENSION @ WELDC E-E

$$f_t = (P/2) / A_t = W / 2A_t$$

$$A_t = 11.94(3.00) = 35.82 \text{ in}^2$$

$$f_t = W (0.01396)$$

$$f_t = \underline{\underline{3972 \text{ PSI}}}$$
  

### SHEAR FOR TEAR-OUT

$$f_v = (P/2) / 2A_v = W / 4A_v$$

$$2A_v = 3(11.94 - 6.028) = 17.736 \text{ in}^2$$

$$f_t = W (0.02819)$$

$$f_t = \underline{\underline{8020 \text{ PSI}}}$$

REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

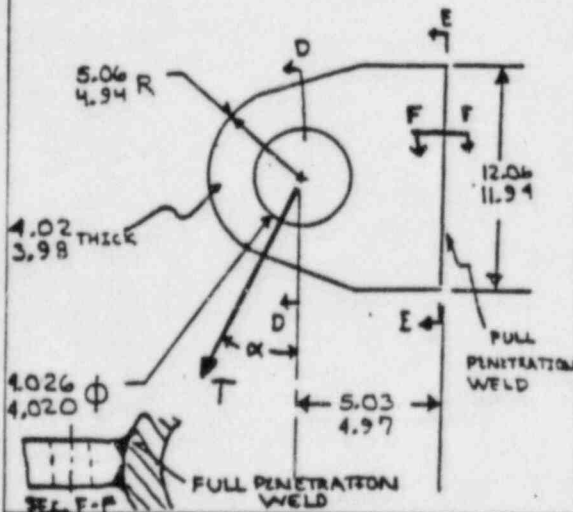
TITLE R. V. Internals Lift Rig				PAGE 20 of 67	
PROJECT PSE/PNJ	AUTHOR <i>John J. ...</i>	DATE 12/82	CHK'D. BY <i>Joe Richard</i>	DATE 2/83	CHK'D. BY DATE
S.O. PTNP-188	CALS. NO.	FILE NO.	GROUP CHE		
SUPPORT PLATE & SUPPORT PIPE <div style="display: inline-block; vertical-align: middle; margin-left: 10px;"> <div style="border: 1px solid black; border-radius: 50%; width: 30px; height: 30px; display: flex; align-items: center; justify-content: center; margin-bottom: 5px;">10</div> <div style="border: 1px solid black; border-radius: 50%; width: 30px; height: 30px; display: flex; align-items: center; justify-content: center;">9</div> </div>			$W = 284,500 \#$ TENSION IN PLATE  $f_t = P/A_t = W/A_t$ $A_t = 19.94$ $f_t = (1.002515)W$ $f_t = \underline{716 \text{ PSI}}$		
MAT'L SUPPORT PIPE ASTM A106  SUPPORT PLATE ASTM A515 GR. 70			TENSION IN PIPE  $f_t = P/A_t = W/A_t$ $A_t = \pi/4(d_o^2 - d_i^2)$ $= \pi/4(17.94^2 - 14.421^2)$ $= 89.44 \text{ in}^2$ $f_t = W(0.011181)$ $f_t = \underline{3181 \text{ PSI}}$		
REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE

WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 21 OF 67	
PROJECT PSE/PNJ	AUTHOR <i>[Signature]</i>	DATE 11/6	CHK'D. BY <i>[Signature]</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

SIDE LUG (II)A

MAT'L  
ASTM A 515 GR 70  
QUENCHED &  
TEMPERED



$W = 284,500 \text{ lb}$   
TENSION @ HOLE

THE AREA WILL CONSERVATIVELY  
BE TAKEN AS THE MINIMUM  
OBTAINED FOR HORIZONTAL FORCES

$$f_t = T/A_t = .418W/A_t$$

$$A_t = \min [4.94(2) - 4.026] * 3.98$$

$$= 23.31 \text{ in}^2$$

$$f_t = W(0.017932)$$

$$f_t = \underline{\underline{5102 \text{ PSI}}}$$

COMBINED STRESS  
@ WELD

TENSION:  $f_t = T \sin \alpha / A_t$   
 $= .418W \sin 37.13^\circ / 11.94(3.98)$   
 $= W(0.005307)$

SHEAR:  $f_v = T \cos \alpha / A_v$   
 $= .418W \cos 37.11^\circ / 11.94(3.98)$

BENDING:  $f_b = M_c / I$   
 $= [T \cos \alpha 5.03] [11.94/2] / [3.98 + 11.94^2/12]$

WHERE  $I = bh^3/12$

$$f_b = (418W)(4.015)(5.970) / 564.57$$

$$f_b = W(0.017731)$$

$$f_y = (0.007014)W$$

$$f_t + f_b = W(0.005307 + 0.017731)$$

$$= W(0.023038)$$

$$f_{comb} = \underline{\underline{6554 \text{ psi}}}$$

$$f_y = W(0.007014)$$

$$f_v = \underline{\underline{1995 \text{ psi}}}$$

BEARING

BEARING IS THE SAME AS THE CLUIDS PINS  
INNER BEARING =  $.02630W = 7481 \text{ PSI}$

REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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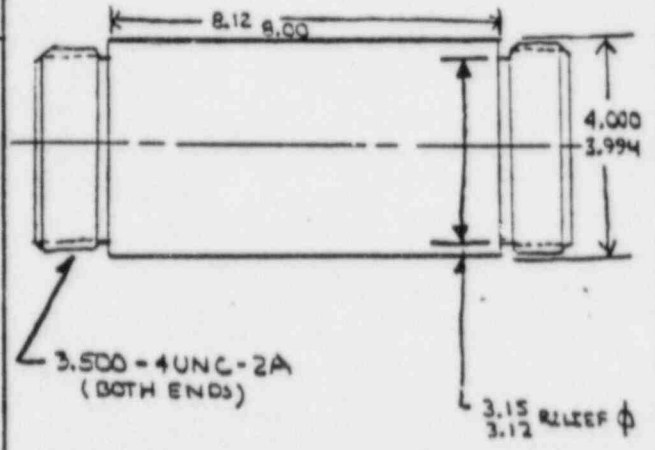
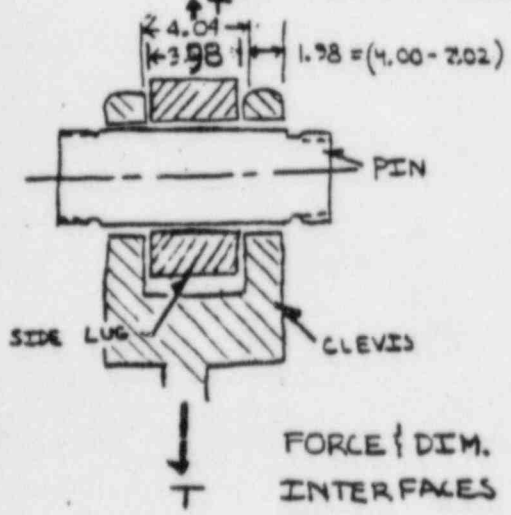


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TITLE R. V. Internals Lift Rig		PAGE 22 of 67	
PROJECT PSE/PNJ	AUTHOR <i>J. S. ...</i>	DATE 12/82	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CALL. NO.	FILE NO.	GROUP CHE

CLEVIS PIN (12) A

EST WT: 40#  
MAT'L: ASTM A434 CLASS BD  
AISI A340 STEEL



W = 284,500#  
SHEAR STRESS'S

$$f_v = P / 2A_v$$

$$P = T = .418W$$

$$A_v = \pi d^2 / 4$$

$$= \pi (3.994)^2 / 4$$

$$= 12.53 \text{ in}^2$$

$$f_v = .418W / (2 * 12.53)$$

$$f_v = (.01668)W$$

$$\underline{f_v = 4745 \text{ psi.}}$$

REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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TITLE R. V. Internals Lift Rig		PAGE 73 of 67	
PROJECT PSE/PNJ	AUTHOR J. Anulan	DATE 12/82	CHK'D. BY J. W. Rusk
S.O. PTNP-188	SCALE NO.	FILE NO.	GROUP CHE

CLEVIS PIN

(12) A

BEARING STRESS

1 PIN TO CLEVIS

$$f_c = (P/2) / A_v$$

$$(P/2) = T/2 = (.418W/2)$$

$$A_v = dt = 3.994(1.98)$$

$$= 7.91 \text{ in}^2$$

$$f_c = .418W / (2 \cdot 7.91)$$

$$f_c = W(.02642)$$

$$f_c = \underline{\underline{7517 \text{ PSI}}}$$

1 PIN TO SIDE LUG

$$f_c = P / A_v$$

$$P = T = .418W$$

$$A_v = dt = 3.994(3.98)$$

$$= 15.896 \text{ in}^2$$

$$f_c = .418W / 15.896$$

$$f_c = W(.02630)$$

$$f_c = \underline{\underline{7481 \text{ PSI}}}$$

BENDING STRESS

$$P = T = .418W$$

$$a = 1.98$$

$$l = 3.98$$

$$g = (4.04 - 3.98) / 2 = 0.03$$

$$d = 3.994$$

$$f_b = (P/2) \left( \frac{a}{2} + g + \frac{d}{4} \right) \frac{32}{\pi d^3}$$

$$f_b = .418W / 2 (1.98/2 + 0.03 + 3.98/4)$$

$$= 32 / (\pi \cdot 3.994^3)$$

$$f_b = W(2.06733)$$

$$f_b = \underline{\underline{19,155 \text{ PSI}}}$$

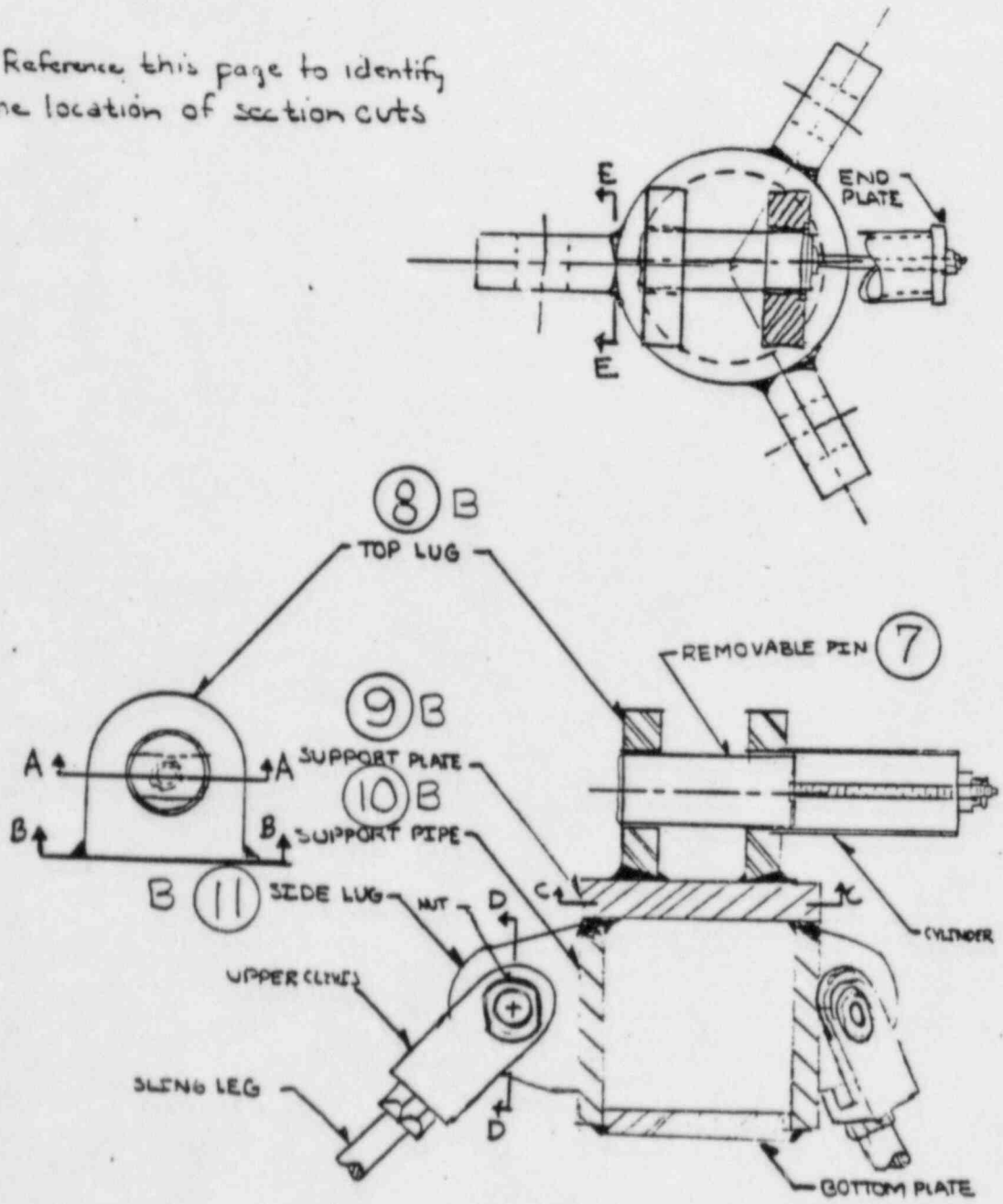
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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 24 OF 67	
PROJECT PSE/PNJ	AUTHOR <i>J. Anderson</i>	DATE 1/26/62	CHK'D. BY <i>JW Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

• BLOCK ASSEMBLY • PNJ

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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 25 of 67	
PROJECT PSE/PNJ	AUTHOR <i>Amundson</i>	DATE 12/62	CHK'D. BY <i>JW Rishard</i>
S.O. PTNP-188	CAUC. NO.	FILE NO.	GROUP CHE

TOP LUG (8)B

BEARING @ A-A

THE BEARING STRESS IS THE SAME AS THE OUTER BEARING STRESS OF THE REMOVABLE PIN. (7)

$$f_t = W (0.02819)$$

$$f_t = \underline{8020 \text{ PSI}}$$

MAT'L  
ASTM A533 GR. B  
CLASS 1  
QUENCHED & TEMPERED

TENSION @ WELDC @ B

$$f_t = (P/2) / A_t = W / 2A_t$$

$$A_t = 11.94(3.00) = 35.82 \text{ in}^2$$

$$f_t = W (0.01396)$$

$$f_t = \underline{3972 \text{ PSI}}$$

TENSION @ A-A

$$f_t = (P/2) / A_t$$

$$P = W$$

$$A_t = 3(11.94 - 6.028)$$

$$= 17.736 \text{ in}^2$$

SHEAR FOR TEAR-OUT

$$f_v = (P/2) / 2A_v = W / 4A_v$$

$$2A_v = 3(11.94 - 6.028) = 17.736 \text{ in}^2$$

$$f_t = W (0.02819)$$

$$f_t = \underline{8020 \text{ PSI}}$$

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TITLE R. V. Internals Lift Rig		PAGE 26 of 67	
PROJECT PSE/PNJ	AUTHOR <i>Amorlan</i>	DATE 12/92	CHK'D. BY <i>W Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

SUPPORT PLATE ↓  
SUPPORT PIPE

(9) B  
(10) B

MAT'L

SUPPORT PIPE

ASTM A106 GRB

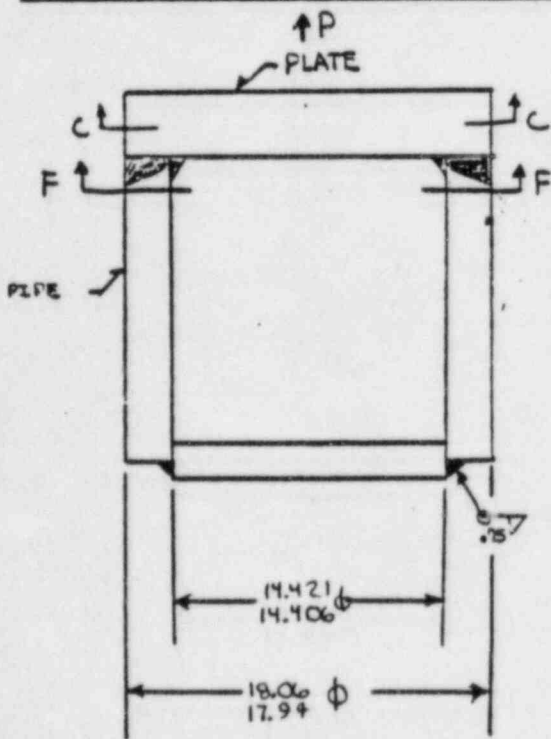
SMLS. STEEL

SUPPORT PLATE

ASTM A 533 GR. B

CLASS I

QUENCHED & TEMPERED



$W = 284,500 \#$   
TENSION IN PLATE

$$f_t = P/A_t = W/A_t$$

$$A_t = \pi 17.94^2/4 = \pi d^2/4$$

$$= 252.8 \text{ in}^2$$

$$f_t = W(0.003956)$$

$$f_t = \underline{1125 \text{ PSI}}$$

TENSION IN PIPE

$$f_t = F/A_t = W/A_t$$

$$A_t = \pi/4(d_o^2 - d_i^2)$$

$$= \pi/4(17.94^2 - 14.421^2)$$

$$= 89.44 \text{ in}^2$$

$$f_t = W(0.01181)$$

$$f_t = \underline{3181 \text{ PSI}}$$

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TITLE R. V. Internals Lift Rig		PAGE 27 of 67	
PROJECT PSE/PNJ	AUTHOR <i>J. Am... 12/82</i>	DATE 12/82	CHK'D. BY <i>JW Birk... 2/83</i>
S.O. PTNP-188	CALE. NO.	FILE NO.	GROUP CHE

SIDE LUG (11) B

MAT'L  
ASTM A588 GR.A  
CLASS 1  
QUENCHED @ 1650°F  
TEMPERED @ 1200°F  
TENSILE STRENGTH -  
80,000 TO 92,000 PSI.

$$f_t = T/A_t = .418W/A_t$$

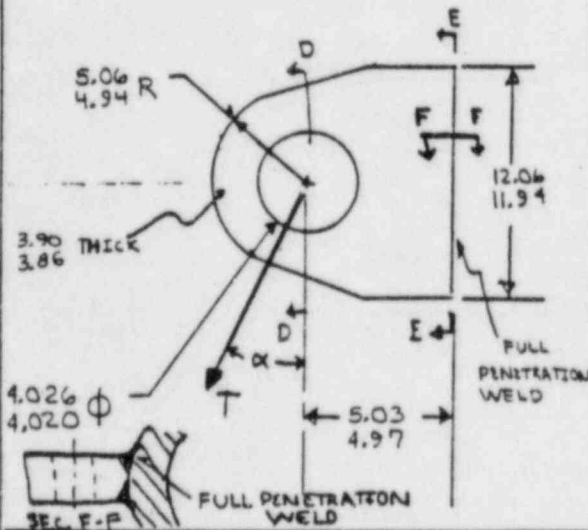
$$A_{t_{min}} = [4.94(2) - 4.026] + 3.86$$

$$= 22.60 \text{ in}^2$$

$$f_t = W(.018498)$$

$$f_t = \underline{5263 \text{ PSI}}$$

COMBINED STRESS @ WELD



TENSION:  $f_t = T \sin \alpha / A_t$   
 $= .418W \sin 37.11^\circ / [1.9 + 3.86]$   
 $= W(.005472)$

SHEAR:  $f_v = T \cos \alpha / A_v$   
 $= .418W \cos 37.11^\circ / [1.9 + 3.86]$

BENDING:  $f_b = M_c / I$   
 $= [T \cos \alpha \cdot 5.03] [11.94/2] / [3.86 \cdot 11.94^3/12]$

WHERE  $I = bh^3/12$

$$f_b = (418W)(4.011)(5.970) / 547.6$$

$$f_b = W(.018280)$$

W = 284,500  
TENSION @ HOLE

$$f_t + f_b = W(.005472 + .018280)$$

$$= W(.02375)$$

$$f_{comb} = 6758 \text{ psi}$$

$$f_v = W(.007233)$$

$$f_v = \underline{2058 \text{ psi}}$$

THE AREA WILL CONSERVATIVELY  
BE TAKEN AS THE MINIMUM  
OBTAINED FOR HORIZONTAL FORCES

BEARING

BEARING IS THE SAME AS THE CLIPS PINS  
INNER BEARING =  $W(.02711) = \underline{7714 \text{ PSI}}$

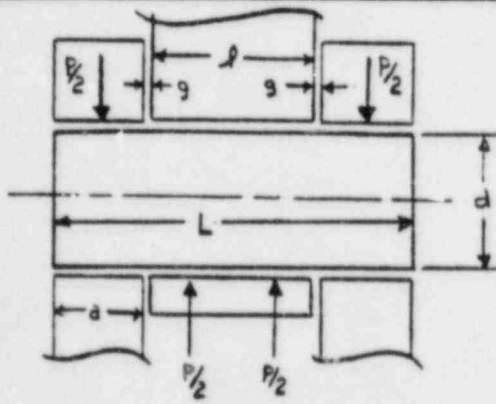
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TITLE R. V. Internals Lift Rig		PAGE 28 OF 67	
PROJECT PSE/PNJ	AUTHOR <i>J. M. ...</i>	DATE 2/83	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CALL. NO.	FILE NO.	GROUP CHE

CLEVIS PIN (12) B

MAT'L: ASTM A434 CLASS BD  
AISI 4340 STEEL



P = force acting on assembly, lb.  
 d = diameter of pin, in.  
 l = length of bearing surface of center body, in.  
 a = length of bearing surface one side of outer body, in.  
 g = gap between bearing surfaces, in.  
 L = total active length of pin, in.  
 =  $l + 2(a + g)$

P = T = .418W lb.  
 d = 3.994 in.  
 l = 3.86 in.  
 a = 1.98 in.  
 g = 0.09 in.  
 =  $(4.04 - 3.86) / 2$

SHEAR STRESS

$$f_v = P / 2A_v$$

$$P = T = .418W$$

$$A_v = \pi d^2 / 4$$

$$A_v = \pi (3.994)^2 / 4$$

$$A_v = 12.529 \text{ in}^2$$

$$f_v = (.418W) / (2 \times 12.529)$$

$$= W(.016682)$$

$$f_v = 4746 \text{ PSI}$$

BEARING STRESS

$$f_c = P / A_v$$

$$P = T = .418W$$

INNER

$$A_v = d l = (3.994)(3.86)$$

$$= (15.417) \text{ in}^2$$

$$f_c = (.418W) / (15.417)$$

$$= W(.02711)$$

$$f_c = 7714 \text{ PSI}$$

OUTER

$$A_v = 2ad = 2(1.98)(3.994)$$

$$= (15.816) \text{ in}^2$$

$$f_c = (.418W) / (15.816)$$

$$= W(.02643)$$

$$f_c = 7519 \text{ PSI}$$

REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE
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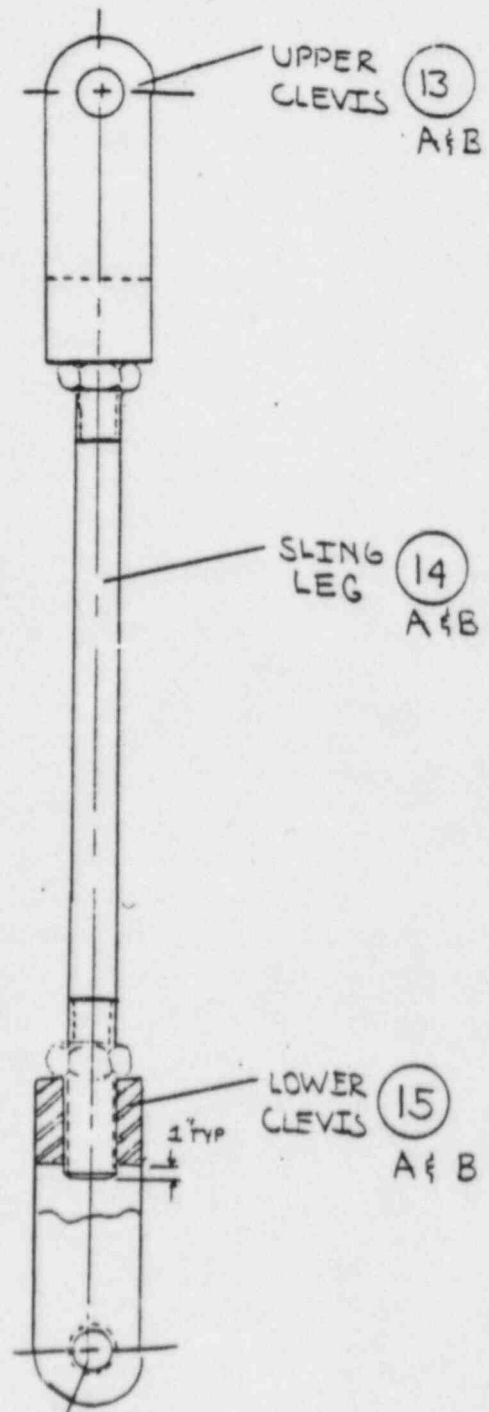
WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig						PAGE 29 OF 67	
PROJECT PSE/PNJ	AUTHOR <i>Amulur</i>	DATE 12/82	CHK'D. BY <i>JW Richard</i>	DATE 2/83	CHK'D. BY	DATE	
S.O. PTNP-188	CAVC. NO.	FILE NO.	GROUP CHE				
<p><b>BENDING STRESS*</b></p> $f_b = (P/2 \times 3/2 + q \times 1/4) \frac{32}{\pi d^3}$ $f_b = (.418W)/2 \times (1.98/2 + .09 + 3.86/4) \times 32 / (\pi (3.994)^3)$ $= W (.06833)$ $f_b = \underline{19,440 \text{ PSI}}$							
REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE

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TITLE R. V. Internals Lift Rig				PAGE 30 of 67	
PROJECT PSE/PNJ	AUTHOR <i>[Signature]</i>	DATE 12/83	CHK'D. BY <i>[Signature]</i>	DATE 2/83	CHK'D. BY <i>[Signature]</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE		

- SLING LEG ASSEMBLY -



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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 31 of 67	
PROJECT PSE/PNJ	AUTHOR <i>Richard</i>	DATE 12/82	CHK'D. BY <i>JW Richard</i>
S.O. PTNP-188	CALL. NO.	FILE NO.	GROUP CHE

UPPER CLEVIS

(13)  
A & B

PSE  
MAT'L: SA-508 CL 2  
QUENCHED AND TEMPERED  
YIELD 50,000 PSI  
TENSILE 80,000 PSI

PNJ  
MAT'L: ASTM A668 AISI 4340 STEEL  
CLASS M

W = 284,500 #  
TENSION @ A-A

$$f_t = P / A_t$$

$$A_t = (7.94 - 4.023)(1.98)$$

$$= 7.756 \text{ in}^2$$

$$P = T/2 = .418 W / 2$$

$$f_t = .418 W / 7.756 / 2$$

$$f_t = W (0.02695)$$

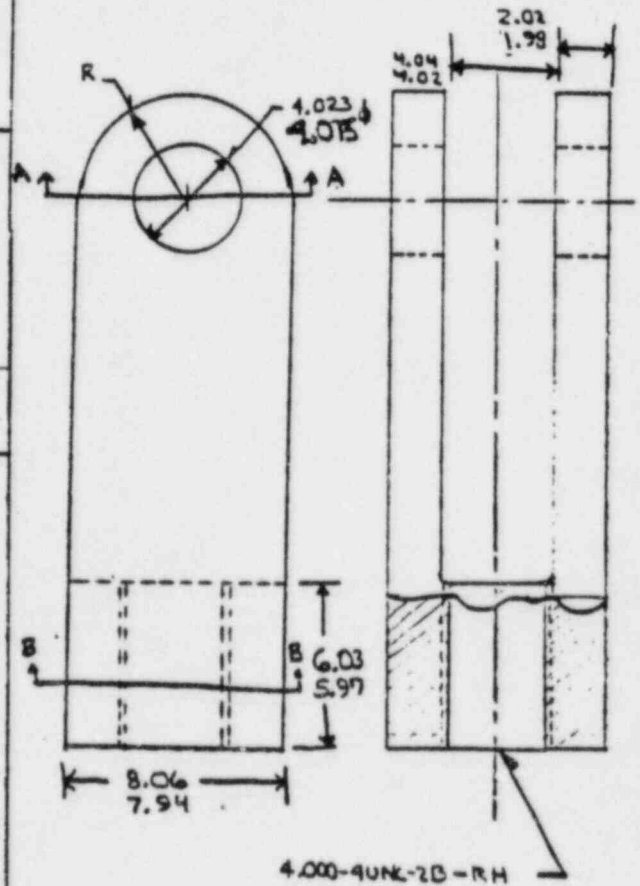
$f_t = 7666 \text{ PSI}$

BEARING @ A-A

BEARING STRESS @ A-A WAS  
PREVIOUSLY CALCULATED DURING  
CLEVIS PIN ANALYSIS.

$$f_c = W (.02642)$$

$f_c = 7527 \text{ PSI}$



THREAD SHEAR

(SEE SLING LEG ANALYSIS THAT FOLLOWS)

$$f_v = W (0.011058)$$

$f_v = 3288 \text{ PSI}$

TEAR-OUT SHEAR

$$f_v = (P/2) / 2A_v = (.418 W / 2) / (2 \cdot [(7.94 - 4.023) / 2] \cdot 1.98) = W (.02695)$$

$$f_v = 7666 \text{ PSI}$$

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S.O. PTNP-188	CALE. NO.	FILE NO.	GROUP CHE

SLING LEG (14) A+B

PSE  
MATL: AISI-1117  
HOT ROLLED  
OR 1018 COLD ROLLED

PNJ  
MAT'L ASTM A434 CL BC  
AISI 4340 TURNED GROUND  
OR  
ASTM A588  
NORMALIZED, TURNED GROUND

W = 284,500 #  
THREAD SHEAR @ A-A

$$f_v = P / A_v$$

$$A_v = \pi D_{pitch} \times l / 2$$

FROM MARK'S HANDBOOK FOR M.E.'S  
3RD ED. P 8-10

$$D_{pitch} = 3.838 \text{ in}$$

FROM CLIVES DWG (ASSY SHOWS THREADS  
EXTEND 1" (TIP) PAST CLIVES THREADING)

$$l = 6.0 \text{ in}$$

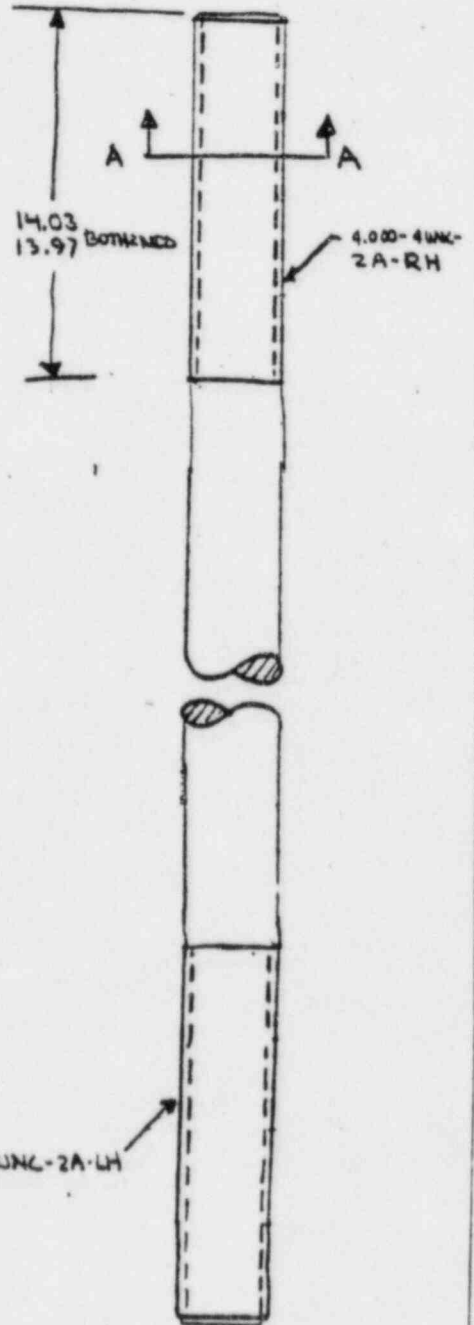
$$A_v = \pi (3.838) 6.0 / 2 = 36.17 \text{ in}^2$$

$$P = T = .418 W$$

$$f_v = .418 W / 36.17$$

$$f_v = W (0.011557)$$

$$f_v = \underline{\underline{3288 \text{ PSI}}}$$



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SLING LEG (14) A <sub>t</sub> B			SLING LEG (14) B		
<p>(PSE) TENSILE STRESS</p> $f_t = P/A_t$ <p>C A-A FROM MARK'S HANDBOOK FOR MECHANICAL ENGINEERS 10-10 8th Ed; <math>A_t = \text{STRESS AREA} = 11.0805 \text{ in}^2</math></p> <p>THEREFORE</p> $f_{t \max} = P/11.0805$ $P = T = 418 \text{ W}$ $f_{t \max} = 418 \text{ W} / 11.0805$ $f_{t \max} = W(0.03772)$ $\underline{f_{t \max} = 10,732 \text{ PSI.}}$			<p>(PNJ) PNJ HAS A THREAD RELIEF</p> $f_t = P/A_t$ $P = T = 418 \text{ W}$ $A_t = \frac{1}{4} (3.617)^2 = 10.275$ $f_t = 418 \text{ W} / 10.275$ $= W(0.04068)$ $\underline{f_t = 11,574 \text{ PSI}}$		
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LOWER CLEVIS

(15)  
A & B

LOWER CLEVIS' DIMENSIONS ARE THE SAME AS THE UPPER CLEVIS' EXCEPT THAT THE UPPER HOLE THROUGH BOTH LUGS IS THREADED WITH 4.000-4UNC-2B-RH AND THE THREAD WHICH ATTACHES TO THE SLING LEG IS LH.

$$W = 284,500 \#$$

THREAD SHEAR

FROM THE SLING LEG CALCULATIONS

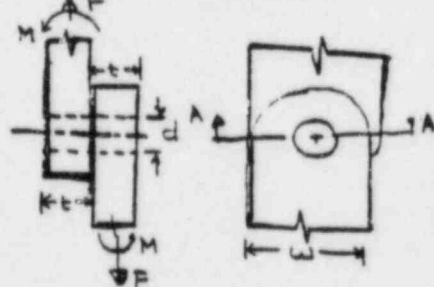
$$f_v = W(0.011557)$$

$$f_v = \underline{3288 \text{ PSI}}$$

TENSION

AT SECTION A-A (SEE UPPER CLEVIS SKETCH) THE FIBER TENSILE STRESS IS FOUND TO BE (FROM SEC 9.5.2 OF TECHNOLOGY INCORPORATED STRESS ANALYSIS MANUAL)

$$f_{t \max} = \frac{F}{(w-d)t} + \frac{GM}{(w-d)t^2}$$



$$F = T/2 = \left(\frac{0.413}{-2}\right)W$$

$$M = F(1.99)/2 = 0.2080W$$

$$f_{t \max} = W \left[ \frac{0.209 + 6(208)/1.99}{(7.94 - 4.023)1.99} \right]$$

$$= W(0.10727)$$

N.B. THE THREADED BOLT REDUCES THE ASSUMED MOMENT.

$$f_{t \max} = \underline{30,518 \text{ PSI}}$$

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Check of extreme fiber stress to AISC Criteria

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From MANUAL OF STEEL CONSTRUCTION, AISC Ed 7., CHAPTER 5

SECTION 1.6.2  
AXIAL TENSION & BENDING  
- FORMULA (1.6-1b) APPLIES

$f_a$  = computed bending tensile stress  
 $f_b$  = computed axial stress  
 $F_y$  = yield stress = 50,000 psi

$F_a$  = axial stress permitted if axial force alone existed  
 = .60  $F_y$  per sec 1.5.1.1  
 $F_b$  = bending stress permitted if bending moment alone existed  
 = .75  $F_y$  per sec 1.5.1.4.3  
 FORMULA (1.6-1b)

$$\frac{f_a}{0.60 F_y} + \frac{f_b}{F_b} + \frac{f_{by}}{F_{by}} \leq 1.0$$

FROM THE PREVIOUS CALCULATIONS

$$f_{by} = 0$$

$$f_{bx} = \frac{6M}{(w-d)t^2}$$

$$= \frac{6(.2080)W}{(7.94-4.023)1.99^2} = .08046W$$

$$= 22,891 \text{ psi}$$

$$f_a = \frac{E}{(w-d)t} = \frac{.418W/2}{(7.94-4.023)1.99} = .02681W$$

$$= 7,628 \text{ psi}$$

$$\frac{f_a}{.60 F_y} + \frac{f_{bx}}{F_{bx}} + \frac{f_{by}}{F_{by}} =$$

$$\frac{7,628}{50,000(.60)} + 0 + \frac{22,891}{50,000(.75)} \quad \text{(A)}$$

$$= .865 \leq 1.0$$

∴ STRESSES ARE BELOW AISC ALLOWABLES

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$$\frac{7,628}{110,000(.60)} + 0 + \frac{22,891}{110,000(.75)} \quad \text{(B)}$$

$$= .393 \leq 1.0$$

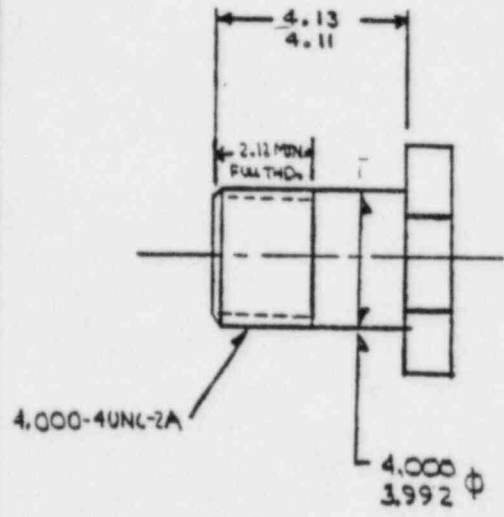
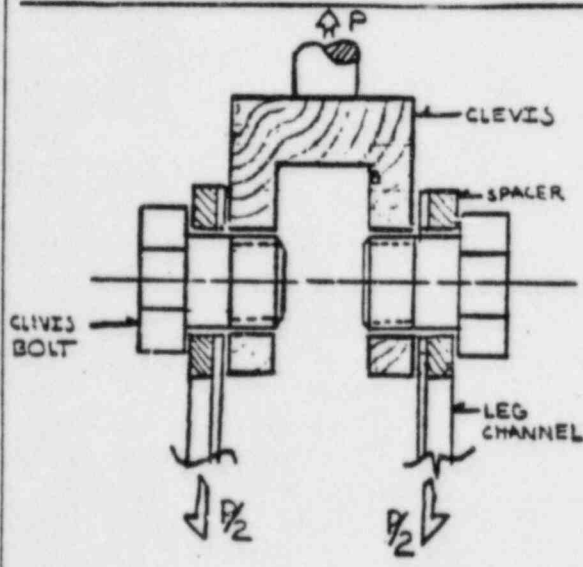
∴ STRESSES ARE BELOW AISC ALLOWABLES

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CLEVIS BOLT (16)



MAT'L:  
ASTM A434  
CLASS BD  
AISI 4340 STEEL

SHEAR

$$f_v = P/2A_v = T/2A_v$$

$$A_v = \pi/4 D_{minor}^2$$

WHERE  $D_{MINOR}$  = ROOT DIAMETER OF THREAD

FOR A 4.000-4UN-2A THD

$$D_{MINOR} = 3.6933 \text{ in}$$

$$T = .418 \text{ W}$$

$$f_v = .418 \text{ W} / (2 * \pi/4 * 3.6933^2)$$

$$= \text{W} (0.01951) \text{ PSI}$$

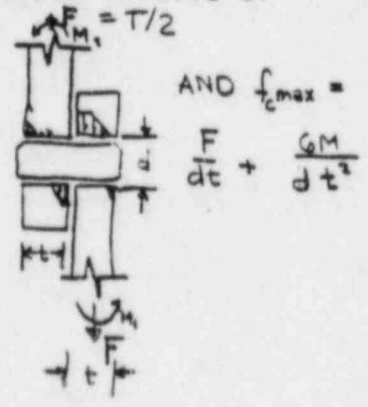
$$f_v = \underline{5551} \text{ PSI}$$

BEARING STRESS

FROM CHAPTER 9, SEC 5 OF TECHNOLOGY/ INCORPORATED'S STRESS ANALYSIS MANUAL:

2) THE AMOUNT OF BENDING IS SIGNIFICANTLY REDUCED BY BOLT CLAMPING OF SINGLE SHEAR JOINTS.

2) ASSUMING AN UNTRADED JOINT THE BEARING STRESS DISTRIBUTION MAY BE MODELED BY



$$f_{c \text{ max}} = \frac{F}{dt} + \frac{GM}{dt^2}$$

$$W = 284,500 \#$$

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CLEVIS BOLT

(16)

$$M = .208W$$

$$F = .209W$$

Assuming even distribution of the moment between the lugs, and assuming yielding is not occurring (checked by  $f_{cmax}$  being  $< F_{YIELD}$ )

$$d = 4.015$$

$$t = 1.99$$

$$M = (F)(1.99)/2 = (P/2)(1.99)/2$$

$$= T(.4975) = W(.208)$$

$$F = T(.500) = .209W$$

$$f_{cmax} = .209W / (4.015 + 1.99)$$

$$+ (6(W(.208) / 4.015 + 1.99^2))$$

$$= W(0.10465)$$

$$f_{cmax} = 29773 \text{ PSI}$$

THIS IS BELOW YIELDING ASSUMPTION, ∴ HOLDS

BENDING STRESS

FROM SEC 9.5.5 THE BENDING STRESS FORMULA FOR THE PIN OF A SINGLE SHEAR JOINT IS

$$f_{bmax} = \frac{1.019 Ft}{d_{pin}^3} \left( \sqrt{\left(\frac{2M}{Ft}\right)^2 + 1} - 1 \right) \frac{2M}{Pt}$$

WITH  $M/Pt = 0.5$

WHEN NOT ALLOWING A PLASTIC ANALYSIS

$$f_b = W(.209) \frac{(10.19)(1.99)}{(3.992)^3} \left[ \left( \frac{2(.208)}{209(1.99)} \right)^2 + 1 \right]^{1/2} - 1$$

$$= W(0.06662)(\sqrt{1.4144} - 1)$$

$$= W(0.02760)$$

$$f_b = 7854 \text{ PSI.}$$

THE BEARING STRESS CALCULATION DONE ON THE PRIOR PAGE IS DONE ONLY TO DETERMINE THE STRESS FIELD, THE ONLY CRITERIA BEING THAT IT BE BELOW YIELD. THE APPROPRIATE CALCULATION FOR ACTUAL BEARING IS THE SAME AS FOR THE SPACER-CHANNEL (21)(20) AND IS

$$f_c = W(.05235)$$

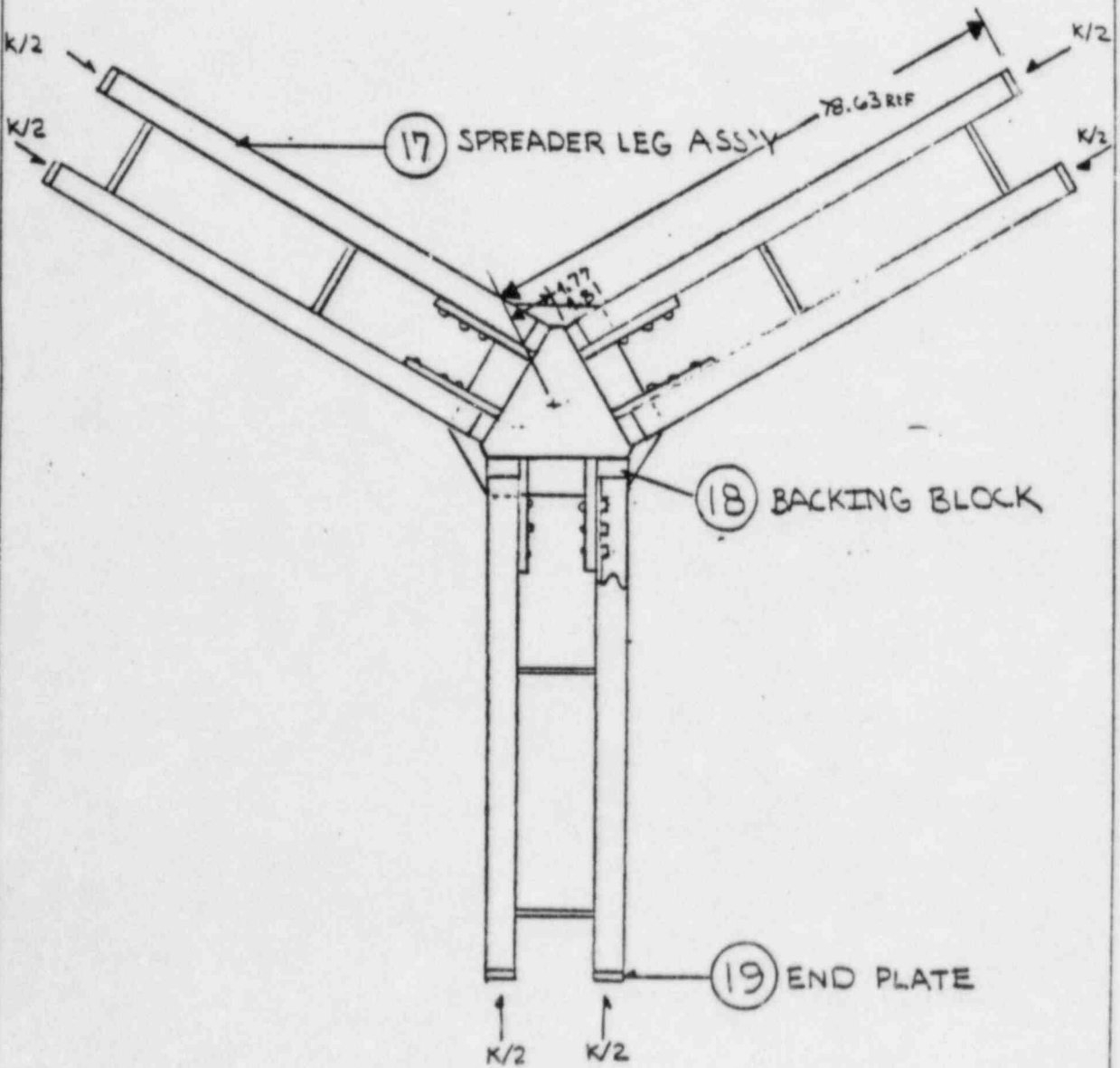
$$f_c = 14,894 \text{ PSI.}$$

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• SPREADER ASSEMBLY •



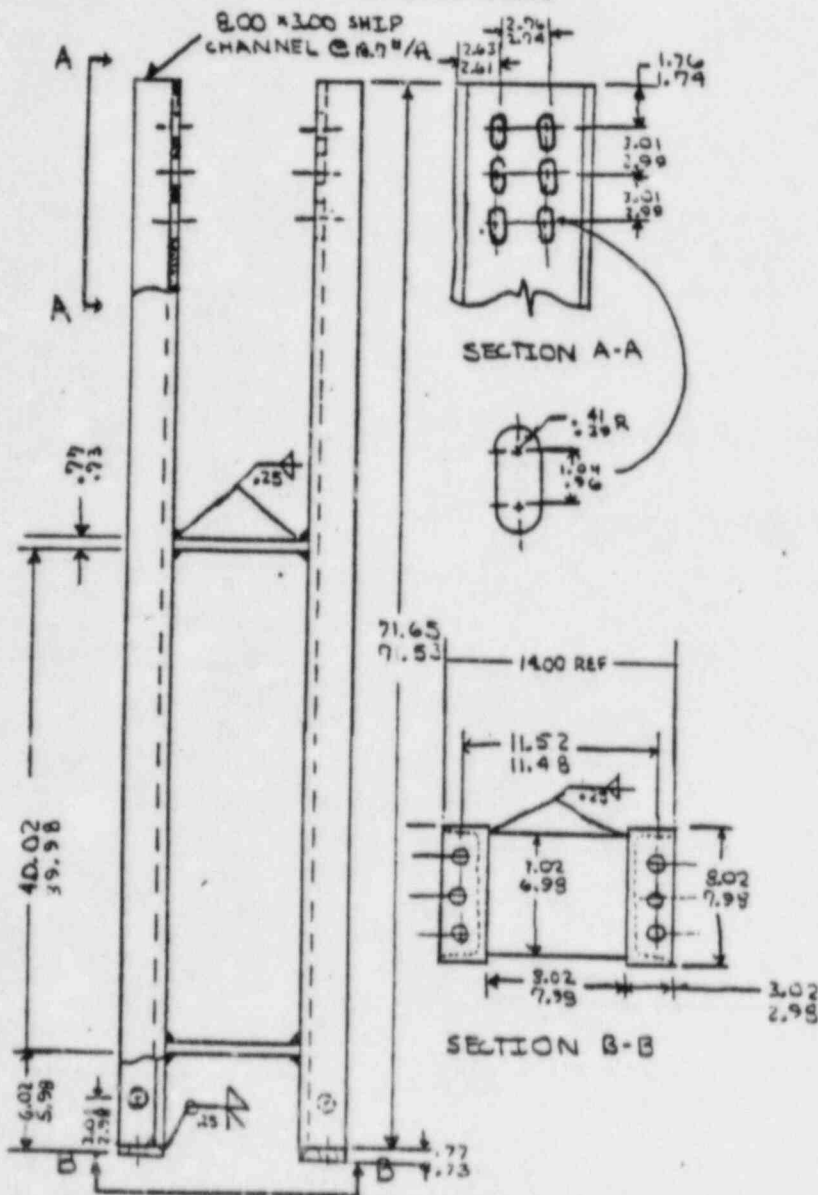
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SPREADER  
LEG ASSEMBLY (17)

MAT'L: ASTM-A36



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SPREADER LEG ASSEMBLY

(17)

$$W = 284,500 \#$$

THE NOMINAL COMPRESSIVE STRESS IN THE SPREADER IS

$$f_t = K/2A_c$$

WHERE K = INWARD FORCE ON SPREADER LEG =  $(.252)W$

$A_c$  = AREA OF ONE CHANNEL OF A LEG

FOR 8x3 @ 18.7 #/A<sub>c</sub> SHIP CHANNEL

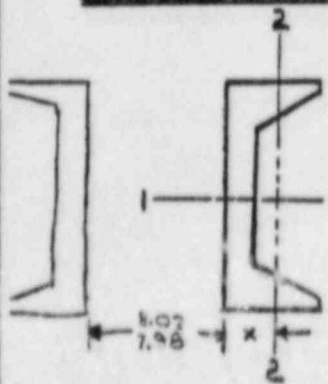
$$A_c = 5.49 \text{ in}^2$$

(FROM MARK'S HANDBOOK FOR MECH. ENGRS P 12-32, 8<sup>th</sup> ED.)

$$f_t = .252W / 2 / 5.49$$

$$f_t = W(0.02295)$$

$$f_t = \underline{6530 \text{ PSI}}$$



$$I_{xx} = 43.7 \text{ in}^4$$

$$r_{x-x} = 2.82 \text{ in}$$

$$S_{xx} = 10.9 \text{ in}^3$$

$$r_{y-y} = 0.60 \text{ in}$$

$$x = 0.57 \text{ in}$$

COMPRESSION IN LEG

FROM THE AISC STEEL CONSTRUCTION MANUAL 7<sup>th</sup> ED., P 5-16 SEC 1.5.1.3

$F_a$  = AXIAL STRESS PERMITTED IN THE ABSENCE OF BENDING

$K_c$  = EFFECTIVE LENGTH FACTOR

$l$  = SPAN LENGTH

$r$  = RADIUS OF GYRATION

$E$  = MODULUS OF ELASTICITY

= 29,000 KIPS/IN<sup>2</sup> FOR STEEL

$F_y$  = MINIMUM YIELD STRESS

$C_c$  = COLUMN SLENDerness RATIO DIVIDING ELASTIC AND INELASTIC BUCKLING

$$C_c = (2\pi^2 E / F_y)^{1/2} = (2\pi^2 29,000,000 / 36,000)^{1/2} = 126.10$$

IF  $K_c l / r < C_c$

ON THE GROSS SECTION OF AXIALLY LOADED MEMBERS

$$F_a = [1 - \frac{1}{2} A'] F_y / (\frac{2}{3} + \frac{1}{8} A' - \frac{1}{8} A'^2)$$

when  $A' = (K_c l / r) / C_c$

IF  $K_c l / r > C_c$

$$F_a = 12\pi^2 E / (23 * (K_c l / r)^2)$$

$$\text{let } K_c = 2.1$$



FROM THE SPREADER ASSEMBLY SKETCH

$$l = 7863 - 4.77 = 73.86$$

$$r = 2(2.82) = 5.64$$

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SPREADER LEG ASSEMBLY (17)

COMPRESSION IN LEG (CONT)

$$I = r^2 A_c$$

$$I = I_{cc} + d^2 A_c$$

d = distance between axis passing through centroid and desired axis

$I_{cc}$  = moment of inertia through centroidal axis

FOR A CHANNEL

$$I_{cc} = r_{cc}^2 A_c = (0.60)^2 5.49 = 1.976 \text{ in}^4$$

$$d = 7.98/2 + .87 = 4.56 \text{ in}$$

$$A_c = 5.49 \text{ in}^2$$

$$I_1 = 1.9769 + (4.56)^2 5.49 = 116.13 \text{ in}^4$$

for two channels

$$I_2 = 232.27$$

$$r_1 = \sqrt{I_2 / (2 \times 5.49)} = 4.60 \text{ in}$$

$$k_1 l / r_1 = 2.1 (7386) / 5.64 = 27.49$$

$$k_2 l / r_2 = 2.1 (73.86) / 4.60 = 33.70$$

FOR BOTH AXES  $k_1 l / r < C_c$

SO

$$F_c = [1 - \frac{1}{2} A'] F_y / (1.3 + \frac{1}{8} A - \frac{1}{8} A')$$

$$A_1 = 27.49 / 126.10 = 0.2180$$

$$A_2 = 33.70 / 126.10 = 0.2672$$

$$F_{c1} = 20,116 \text{ PSI}$$

$$F_{c2} = 19,674 \text{ PSI}$$

THESE ASSUME THE COLUMNS ARE TOGETHER. THE LONGEST UNBRAID LENGTH ABOUT AXIS 2-2 IS 43.02

$$K_c = 0.65^* (\text{FROM 8-138 OF AISC HANDBOOK})$$

$$l = 43.02$$

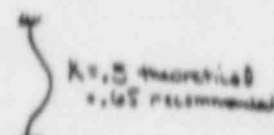
$$r = 0.60$$

$$k_2 l / r = 43.36 < C_c = 126.10$$

$$(k_2 l / r) / C_c = 43.36 / 126.10 = .3439$$

$$F_{c2} = 18,917 \text{ PSI}$$

\*  $K_c$  IS TAKEN AS 2/3 OF THE FULL LENGTH AT THE ENDS WOULD AND (CONSERVATIVELY) MAY DEFLECT SLIGHTLY IF LOOSE. THE CENTER OF THE UNBRAID LENGTH HAS  $K = .65$  AS IT IS WELDED TO THE PLATE AND WOULD BEHAVE LIKE THE MODEL



MAXIMUM BEARING STRESS SAME AS FOR THE BACKING BLOCKS (18)

$$f_c = W (.04124) = 11,734 \text{ PSI}$$

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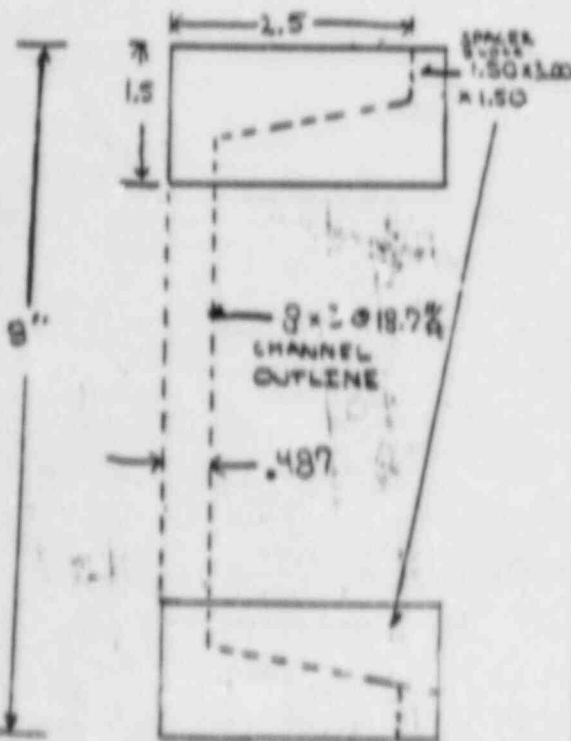
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(18)  
BACKING BLOCK

MAT'L:  
ASTM A 276 TYPE 304  
C.F. COND A.

THE BOLT HOLES ARE  
SLOTTED SO THAT THE  
BACKING BLOCKS RESIST  
THE INWARD FORCE  
K.



BEARING

$$A_{\text{CHANNEL}} = 5.49 \text{ in}^2$$

$$A_{\text{CONTACT}} = A_{\text{CHANNEL}} - [8 - 2(1.5)](.487)$$

$$= 3.055 \text{ in}^2$$

$$P = K/2 = .252W/2$$

$$= .126W$$

$$f_c = P/A_{\text{CONTACT}}$$

$$= .126W / 3.055$$

$$= W(.04124)$$

$$f_c = 11,734 \text{ psi}$$

COMPRESSION IN BLOCK

$$f_c = P/A_b$$

$$P = K/2 = .752W/2$$

$$= .1260W$$

$$A_b = 1.5 \times 3.0 = 4.5$$

$$f_c = .126W/4.5$$

$$= W(.028)$$

$$f_c = 7966 \text{ psi}$$

$$W = 284,500 \#$$

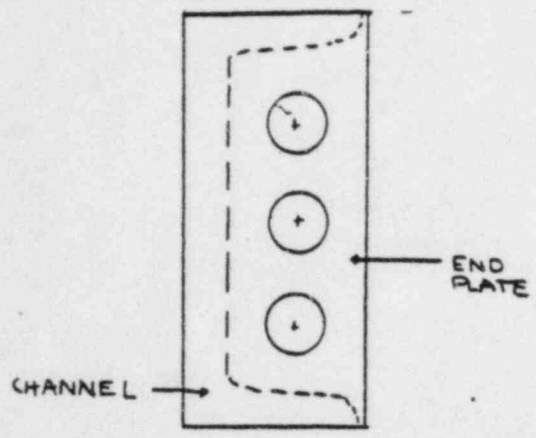
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END PLATE (19)

ASTM - A 36



BEARING  
OF CHANNEL ONTO  
END PLATE (NEGLECTING WELD AREA)

$$f_c = P/A_c$$

$$A_c = A_{\text{CHANNEL}} = 5.49 \text{ in}^2$$

$$P = K/2 = .252 W/2$$

$$f_c = .252 * W/2 / 5.49$$

$$= W(.02295)$$

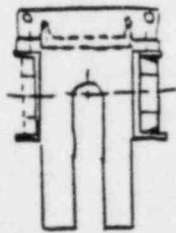
$$f_c = \underline{\underline{6529 \text{ psi}}}$$

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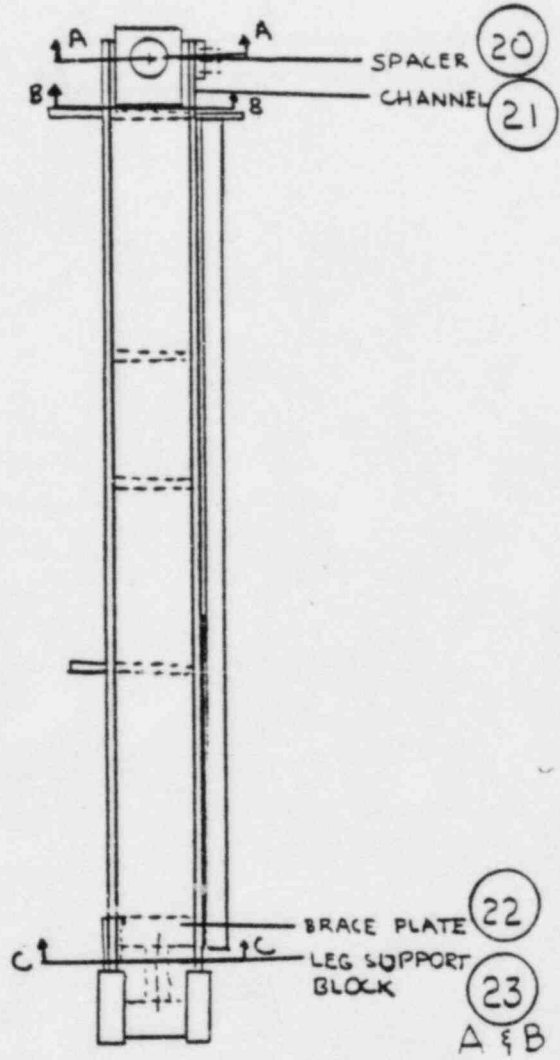
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LEG ASSEMBLY



TOP VIEW



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WESTINGHOUSE NUCLEAR TECHNOLOGY DIVISION

TITLE R. V. Internals Lift Rig		PAGE 45 of 67	
PROJECT PSE/PNJ	AUTHOR <i>John Wilson</i>	DATE 12/62	CHK'D. BY <i>JW Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

SPACER CHANNEL (20/21)

MAT'L  
SPACER - ASTM A588  
GRADES A OR B  
CHANNEL - ASTM-36

$W = 284,500 \#$

BEARING STRESS

THE FIRST BEARING CALCULATION FOR THE CLEVIS BOLT IS A LOCAL FIBER STRESS BEARING WHICH IS CALCULATED MERELY TO PRODUCE A STARTING POINT FOR DETERMINING THE STRESS DISTRIBUTIONS IN THE CLEVIS LUG AND SPACER CHANNEL WELDMENT. TRADITIONALLY, BEARING OF PIN JOINTS IS DONE IGNORING SUCH LOCAL CONCENTRATIONS AND SO THE APPROPRIATE CALCULATION FOR WHICH REFERENCE ALLOWABLES ARE OBTAINABLE IS

$$f_c = P/A_t$$

$$P = T = .4180W$$

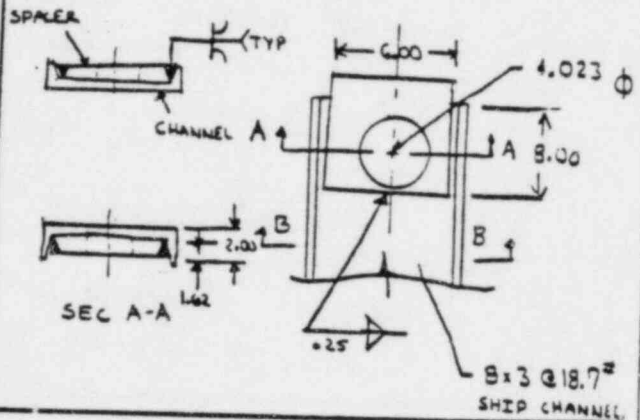
$$A_t = d_t = 3.992(2.00)$$

$$f_c = .4180W / 3.992(2.00)$$

$$= W(.05235)$$

$$f_c = 14,894 \text{ PSI}$$

$W = 284,500 \#$



THE SHEAR IN THE  $\overline{K}$  WELD AND  $\overline{0.25}$  FILLET WELD WOULD BE

$$f_v = P/A_v$$

$$P = (W/3)$$

$$A_v = 8.00(1.62)(2)(2) + .707(6)(2)(.187)$$

$$= 53.96 \text{ in}^2$$

$$f_v = W(0.006177)$$

$f_v = 1757 \text{ PSI}$

IF AS A WORST CASE IT IS ALLOWED THAT ALL FORCES ACT ON THE SPACER

TENSION @ B-B

$$f_t = P/A_t$$

$$P = W/3$$

$$A_t = 2 A_{CHANNEL} = 2(5.49) = 10.98 \text{ in}^2$$

$$f_t = W(0.030358)$$

$$f_t = 8640 \text{ PSI}$$

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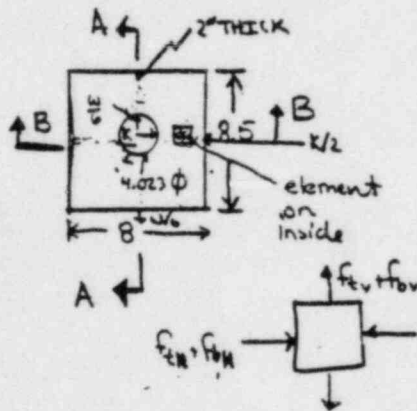
CHECK OF MAXIMUM STRESS TO AISC CRITERIA

$$W = 284,500 \#$$

$$T = .418 W$$

$$K = .252 W$$

THE TENSILE STRESS WILL BE DETERMINED BY SPLITTING THE FORCE T AND THE MOMENTS IT PRODUCES INTO COMPONENTS AND APPROXIMATING THE SPACER-CHANNEL WELDMENT BY THE FOLLOWING:



At the stressed element shown four forces act

$$f_{t \text{ VERTICAL}} = \frac{W}{6} / (8 - 4.023) / 2 = 5,961 \text{ PSI}$$

$$f_{t \text{ HORIZONTAL}} = \frac{K}{2} / (8.5 * 2) = 2,109 \text{ PSI}$$

$$f_{bV} = \frac{M_V C}{I}$$

$$= \left[ \frac{W}{3} / 2 \right] [1] / \left[ \frac{(8 - 4.023)^2}{12} \right]$$

$$= 17,886 \text{ PSI}$$

$$f_{bH} = \frac{M_H C}{I}$$

$$= \left[ \frac{W(.252)}{2} \right] [1] \left[ \frac{8.5^2}{12} \right]$$

$$= 6,326 \text{ PSI}$$

Combining by the square-root-of-the-sum-of-the-squares

$$f_t = \sqrt{f_{tH}^2 + f_{tV}^2} = 6,325 \text{ PSI}$$

$$f_b = \sqrt{f_{bH}^2 + f_{bV}^2} = 18,972 \text{ PSI}$$

THE METHOD FOLLOWED IS SIMILAR TO THAT DONE FOR THE CLEVIS, AND TAKES NO CREDIT FOR THE JOINT BEING BOLTED

USING THE WEAKEST YIELD ( $F_y = 36 \text{ KSI}$ ), AND SO CONSERVATIVELY NEGLECTING THE HIGHER STRENGTH PROVIDED BY THE SPACER ( $F_y = 50 \text{ KSI}$ )

formula (1.6-1b) of chapter 5  
 sec (1.6-1b) of AISC becomes  $\frac{f_t}{.6(36,000)} + \frac{f_b}{.75(36,000)} \leq 1$

$$= .995$$

$$f_{t \text{ max}} = f_b + f_t = 25,297 \text{ PSI}$$

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S.O. PTNP-188	CALL. NO.	FILE NO.	GROUP CHE

LEG SUPPORT BLOCK (23) A  
AND BRACE PLATE (22)

MAT'L  
LEG SUPPORT BLOCK: AISI 8620  
BRACE PLATE : ASTM A-36

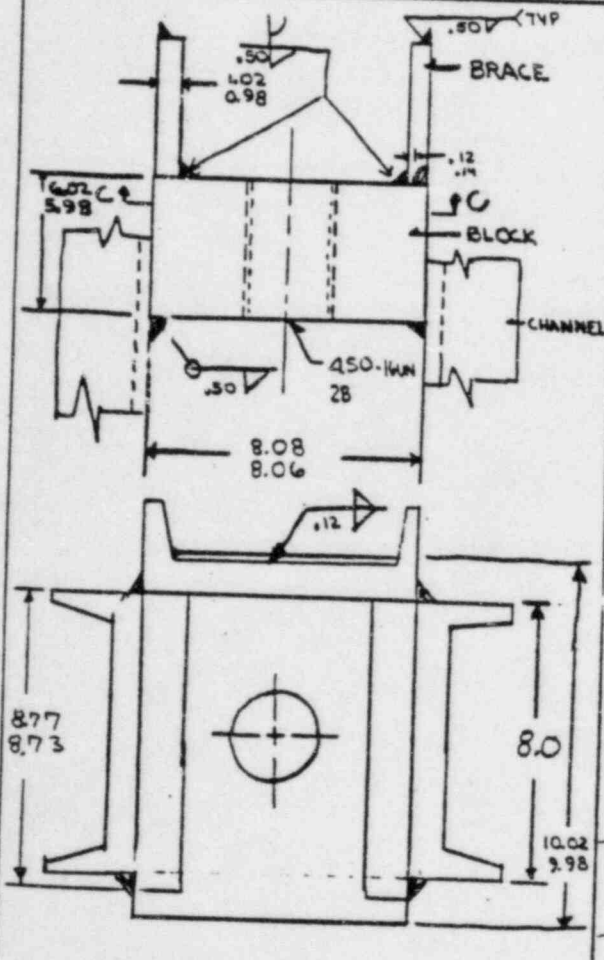
THREAD SHEAR (A)

$$W = 294,500 \#$$

THE THREAD SHEAR IS THE SAME AS THAT FOR THE ADAPTOR

$$f_v = W(0.019879)$$

$$f_v = 5656 \text{ PSI}$$



WELD SHEAR

THE SMALL WELD (.12) BRACING THE BACK CHANNEL - RIFENER HAS NEGLIGIBLE EFFECT ON THE WELD DISTRIBUTION, THE WELD GIVES MORE THAN FULL STRENGTH TO THE BLOCK-PLATE CONNECTION. SO THE SHEAR STRESS IN THE FILLET WELDS IS

$$f_v = P/A_w$$

$$P = W/3$$

$$A_w = (.707)(.50)[8.0 \times 4 + 4(6)]$$

$$= .707 t_{\text{THROAT}} l_{\text{WELD}}$$

$$= 19.796 \text{ in}^2$$

$$f_v = W(0.016838)$$

$$f_v = 4,791 \text{ PSI}$$

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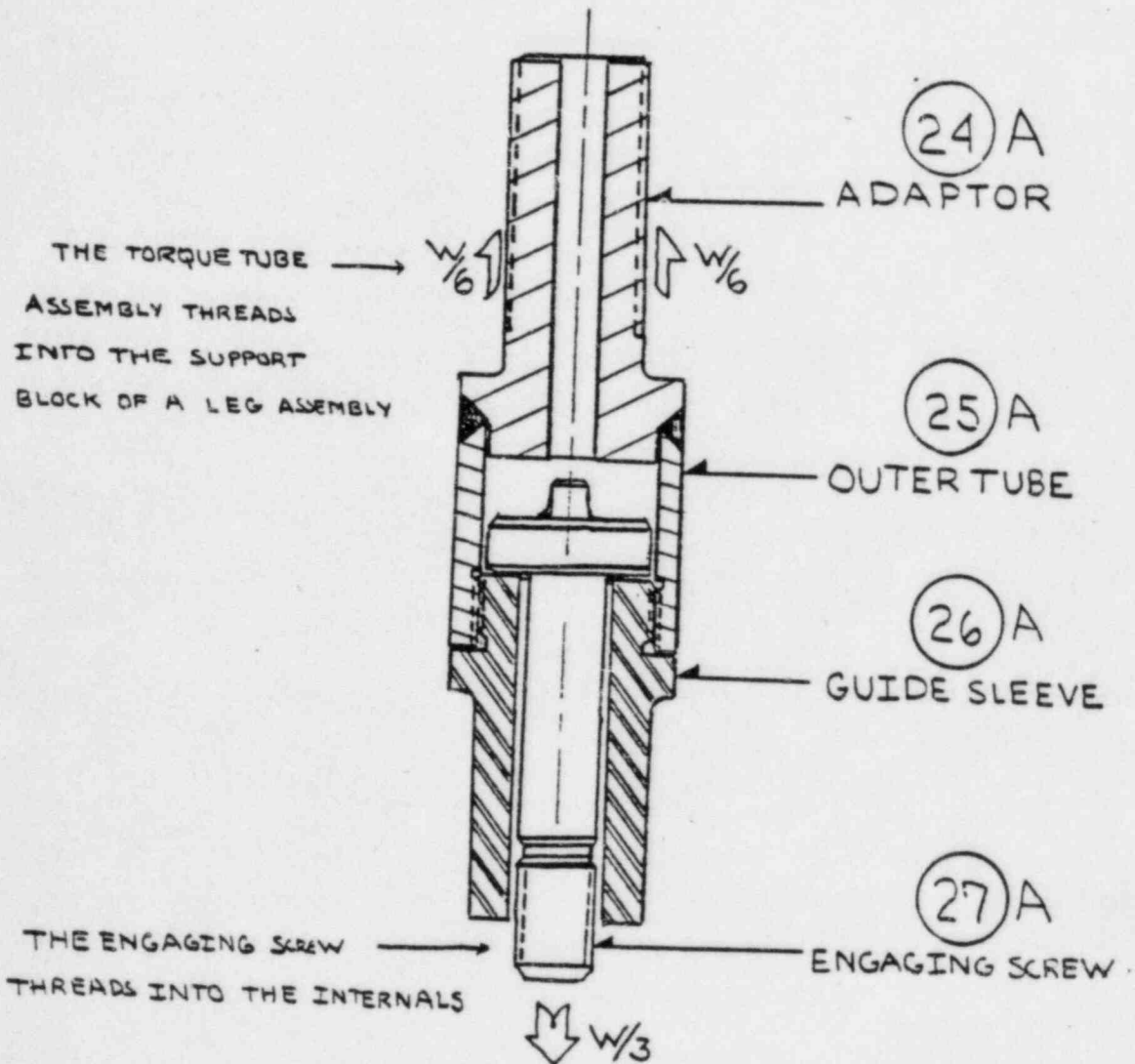
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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE		
LEG SUPPORT BLOCK (23) B					
<p>THE PNJ LEG SUPPORT BRACKET IS THE SAME AS THE PSE UNIT (FOR ANALYSIS) EXCEPT THAT THE 4.50-16UN-2B THREADED HOLE IS REPLACED BY A SMOOTH-BORED HOLE WHOSE DIAMETER IS</p> <p><math>\phi = 4.81</math> 4.75</p>					
<p>BEARING OF NUT (ITEM 28) ON LEG SUPPORT BLOCK</p> $f_c = F/A_c$ $F = W/3$ $A_c = \{ [6.49 - 2(.21)]^2 \cdot 4.81^2 \} \frac{\pi}{4}$ $= 10.767 \text{ in}^2$ $f_c = W/3/10.767$ $= W(.030959)$ $\underline{\underline{f_c = 8,808 \text{ PSI}}}$					
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S.O. PTNP-188	CALC. NO.	FILE NO.	DATE 2/83
		GRO' P CHE	

**TORQUE TUBE ASSEMBLY PSE**



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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

ADAPTOR & OUTER TUBE

24/25 A

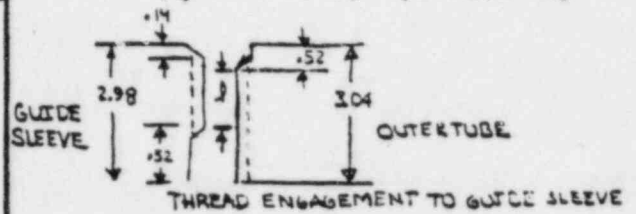
MAT'L:

ADAPTOR - ASTM A-276  
TYPE 304  
HOT ROLLED & PICKLED  
COND. A.

OUTER TUBE - ASTM A-312  
TYPE 304  
SMLS. CF. & HT. TR.

$W = 284,500 \text{ lb}$   
5.5-4 THREAD SHEAR @ 5.5-4 UN

$$f_v = P/A_v = (W/3)/A_v$$



$$l = (3.04 - .52) - .52 = l = 2.00$$

$$A_v = \pi D_{pitch} l / 2$$

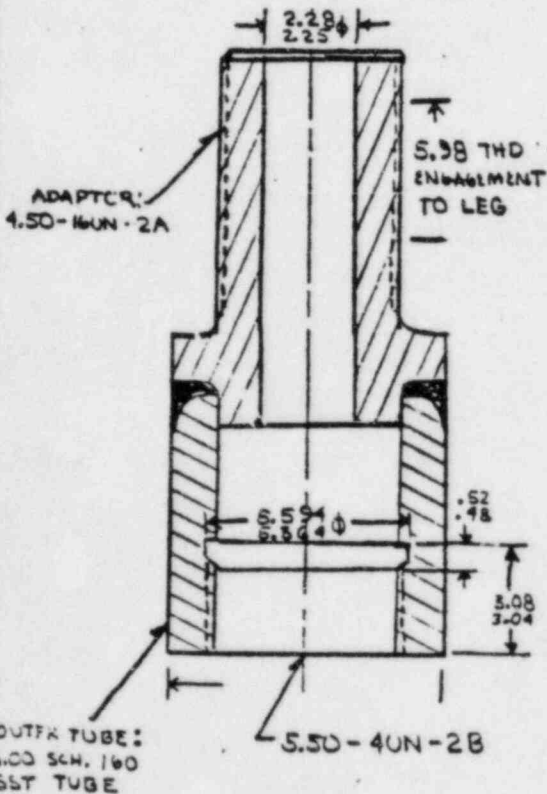
$$D_{pitch} = D_3 - .64952/n$$

$$= 5.5 - .64952/4 = 5.33762$$

$$A_v = \pi (5.33762)(2.00)/2 = 16.769$$

$$f_v = W(0.19879)$$

$$f_v = \underline{5656 \text{ PSI}}$$



THREAD SHEAR @ 4 1/2-16 THD

$$f_v = P/A_v = (W/3)/A_v$$

$$A_v = \pi D_{pitch} l / 2$$

$$l = 5.98 \text{ in}$$

$$D_{pitch} = 4.4594 \text{ (MARKS, 8th Ed)}$$

$$A_v = \pi (4.4594)(5.98)/2 = 41.889 \text{ in}^2$$

$$f_v = W/(3 * 41.889)$$

$$f_v = W(0.007958)$$

$$f_v = \underline{2264 \text{ PSI}}$$

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S.O. PTNP-188	CAUSE NO.	FILE NO.	GROUP CHE		

ADAPTOR &  
OUTER TUBE

(24/25) A

TENSION @ THREAD RELIEF

FROM MARK'S HANDBOOK, P 8-158, 8th ed  
6.00 SCH 160 SST PIPE HAS AN  
O.D. OF 6.625 in.

$$f_t = P/A_t = (W/3)/A_t$$

$$A_t = \pi/4 (d_o^2 - d_i^2)$$

$$= \pi/4 (6.625^2 - 5.594^2)$$

$$= 9.894 \text{ in}^2$$

$$f_t = W(0.03369)$$

$$f_t = \underline{\underline{9585 \text{ PSI}}}$$

TENSION @ 4 1/2 THREAD

FROM MARK'S HANDBOOK, 8th ed  
THE STRESS AREA OF A 4 1/2-16UN-2A  
THREAD IS  $A_s = 15.4662 \text{ in}^2$

$$f_t = P/A_t = (W/3)/A_t$$

$$A_t = A_s - \pi/4 d_i^2$$

$$= 15.4662 - (\pi/4)(2.29)^2$$

$$= 11.383 \text{ in}^2$$

$$f_t = W(0.02928)$$

$$f_t = \underline{\underline{8330 \text{ PSI}}}$$

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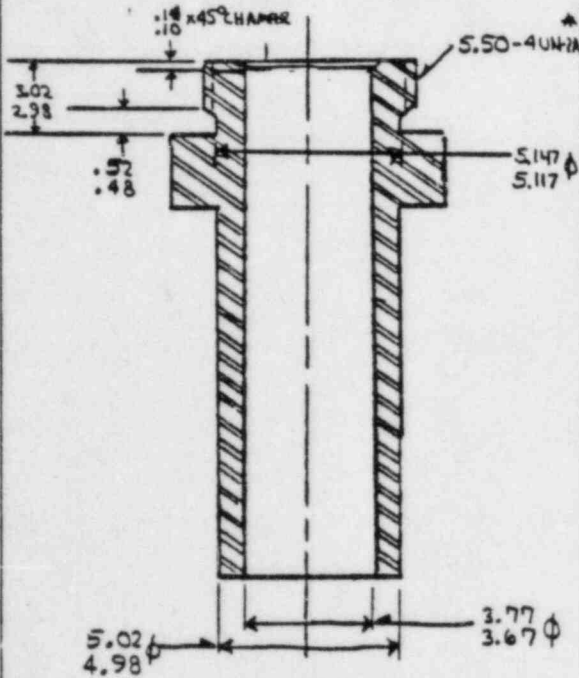


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S.O. PTNP-188	SCALE NO.	FILE NO.	GROUP CHE

GUIDE SLEEVE (26) A

MAT'L: ASTM A 276  
TYPE 304  
HOT ROLLED & PICKLED  
COND. A.



FOR A 5 1/2 - 4UN - 2A THREAD  
 $D_{pitch} = 5.500 - 0.64952/4$   
 $= D_s - 0.64952/n$   
 where  $D_s$  = major diameter and  $n$  = number of threads per inch

$W = 284,500 \#$   
BEARING STRESS

THE BEARING STRESS BETWEEN THE GUIDE SLEEVE AND ENGAGING SCREW IS CALCULATED IN THE ENGAGING SCREW SECTION.

$$f_c = W (0.05961)$$

$$f_c = \underline{16,958 \text{ PSI}}$$

COMPRESSION AT THE RELIEF - AN UPPER LIMIT ANALYSIS -

$$f_t = (-1) P/A_t = (W/3)/A_t$$

$$A_t = \pi/4 (d_o^2 - d_i^2) = \pi/4 (5.117^2 - 3.77^2)$$

$$f_t = -W/(3 * 9.402)$$

$$f_t = -W (0.03545)$$

$$f_t = \underline{-10,086 \text{ PSI}}$$

THREAD SHEAR

$$f_v = P/A_v = (W/3)/A_v$$

$$A_v = D_{pitch} \pi l / 2$$

FROM ADAPTOR/OUTER TUBE  $l = 2.00$

$$A_v = 5.3376 \pi (2.00) / 2 = 16.769$$

$$f_v = W / (3 * 16.769)$$

$$f_v = W (0.019879)$$

$$f_v = \underline{5656 \text{ PSI}}$$

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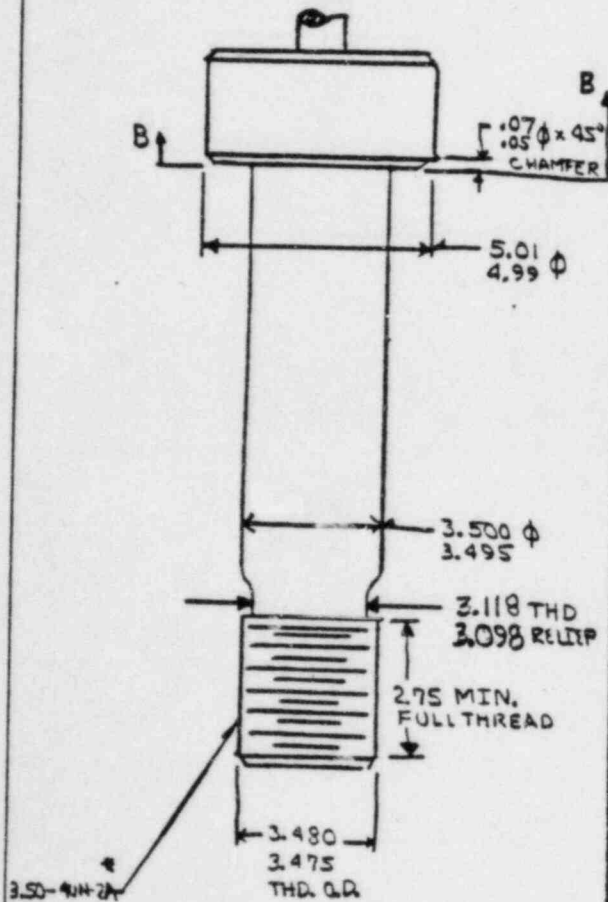
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ENGAGING SCREW

(27) A

MAT'L:  
ASTM A-276  
TYPE 304  
COND. A.



$W = 284,500 \#$   
BEARING @ B-B

$$f_c = P/A_c$$

$$P = (W/3)$$

$$A_c = (d_o^2 - d_i^2) \pi/4$$

$$= (\pi/4) ((4.99 - 2(.07))^2 - (3.77 + 2(.14))^2)$$

$$= 5.592 \text{ in}^2$$

$$f_c = W/(3 \times 5.592)$$

$$f_c = W(0.05961)$$

$f_c = 16,959 \text{ PSI}$

TENSILE @ THREAD RELIEF

$$f_t = P/A_t$$

$$P = W/3$$

$$A_t = \pi/4 d^2 = 3.098^2 (\pi/4) = 7.5379 \text{ in}^2$$

$$f_t = W/(3 \times 7.5379) = W(0.04422)$$

$f_t = 12,581 \text{ PSI}$

THREAD SHEAR

$$f_v = P/A_v = (W/3)/A_v$$

$$A_v = D_{pitch} \pi D/2 = 3.3373 \pi 2.75/2$$

$$f_v = W(0.02312)$$

$f_v = 6577 \text{ PSI}$

\* FROM MARK'S HANDBOOK FOR M.E.S., 8th Ed  
Sec 8:  $D_{pitch} = 3.3373 \text{ in.}$

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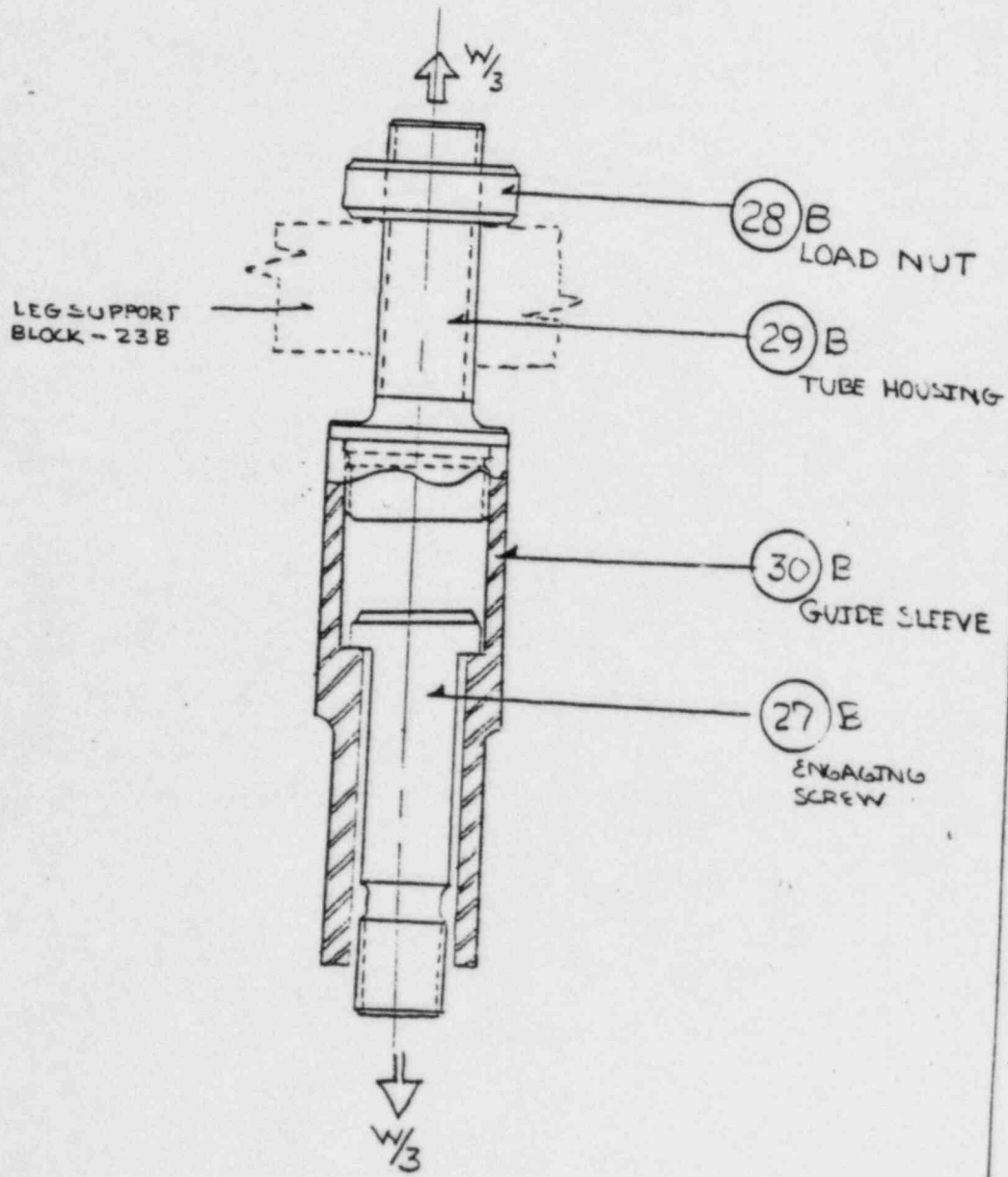
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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE				
<p>ENGAGING SCREW (27) B</p> <p>THE PNJ ENGAGING SCREW IS THE SAME AS THE PSE ENGAGING SCREW EXCEPT THAT THE BEARING STRESS IS AS SHOWN AND THE MATERIAL IS DIFFERENT</p> <p>BEARING ON GUIDE SLEEVE THE BEARING IS THE SAME AS CALCULATED FOR THE GUIDE SLEEVE ITEM 30-B</p> $f_c = W(.04348)$ $f_c = \underline{\underline{12,370 \text{ PSI}}}$ <p>MAT'L ASTM 12-504 GRADE 1030 COND 1100</p>							
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GROUP CHE			

TORQUE TUBE ASSEMBLY - PNJ



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PROJECT PSE/PNJ	AUTHOR <i>J. Amulaw</i>	DATE 12/62	CHK'D. BY <i>JO Bickel</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE
LOAD NUT (28) B			
MAT'L: ASTM A-276 TYPE 304 HOT ROLLED & PICKLED COND A			
BEARING THE BEARING STRESS HAS BEEN CALCULATED IN THE LEG SUPPORT BLOCK CALCS FOR ITEM 23 B  $f_c = W(.030959)$ $f_c = \underline{\underline{8,809 \text{ PSI}}}$			
THREAD SHEAR  $f_v = P/A_v$ $P = W/3$ $A_v = D_p \pi l / 2$ $l = 2.97 \text{ in}$ $D_p = D_s - .64952/n$ $= 4.50 - .64952/4$ $= 4.3376 \text{ in}$ $A_v = 4.3376 (2.97) \pi / 2$ $= 20.24 \text{ in}^2$ $f_v = W/3 / 20.24$ $= W(.016472)$ $f_v = \underline{\underline{4686 \text{ PSI}}}$			
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S.O. PTNP-188	CALC. NO.	FILE NO.	DATE 2/83
		GROUP CHE	

<p style="text-align: center;">TUBE HOUSING (29) B</p> <p>MAT'L: ASTM A-276 TYPE 304 HOT ROLLED &amp; PICKLED COND. A.</p>	<p>THREAD SHEAR ON LOAD NUT CALCULATED AT LOAD NUT ITEM 28-B</p> $f_v = W(.016472)$ $f_v = \underline{\underline{4686 \text{ PSI}}}$
	<p>TENSION AT MINIMUM SECTION</p> $f_t = P/A_t$ $P = W/3$ $A_t = A_T - \pi/4(2.15)^2$ $A_T = \pi/4(D_o - .9743/n)^2$ $= \pi/4(4.5 - .9743/4)^2$ $= 14.229 \text{ in}^2$ $A_t = 14.229 - \pi/4(2.15)^2$ $= 10.599 \text{ in}^2$ $f_t = W/3/(10.599)$ $= W(.031450)$ $f_t = \underline{\underline{8948 \text{ PSI}}}$
	<p>THREAD SHEAR ON GUIDE SLEEVE</p> $f_v = P/A_v = (W/3)/A_v$ $A_v = \pi D_p l/2$ $D_p = D_o - .64952/n =$ $= 5.500 - .64952/4 = 5.3376 \text{ in}$ $l = 3.48 - .53 - .15 = 2.800 \text{ in}$ $A_v = \pi(5.3376)(2.800)/2 = 23.476 \text{ in}^2$ $f_v = W/3/23.476$ $= W(.014199)$ $f_v = \underline{\underline{4040 \text{ PSI}}}$

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S.O. PTNP-188	CALE. NO.	FILE NO.	GROUP CHE				
<p>GUIDE SLEEVE (30) B</p> <p>MAT'L: ASTM A-276 TYPE 304 HOT ROLLED &amp; PICKLED COND A</p>		<p>THREAD SHEAR ON TUBE HOUSING CALCULATED AT TUBE HOUSING CALCS, ITEM 29-B</p> $f_v = W (.014199)$ $f_v = \underline{\underline{4040 \text{ PSI}}}$					
		<p>TENSION AT THREAD RELIEF</p> $f_t = P / A_t$ $P = W / 3$ $A_t = \frac{\pi}{4} (d_o^2 - d_i^2)$ $= \frac{\pi}{4} (6.56^2 - 5.594^2)$ $= 9.221 \text{ in}^2$ $f_t = W / 3 / 9.221$ $= W (.03615)$ $f_t = \underline{\underline{10,284 \text{ PSI}}}$					
		<p>BEARING ON ENGAGING SCREW</p> $f_c = P / A_c$ $P = W / 3$ $A_c = \frac{\pi}{4} (d_o^2 - d_i^2)$ $d_o = 4.99 - 2(.07) = 4.85$ $d_i = 3.682 + 2(.15) = 3.982$ $A_c = 7.67 \text{ in}^2$ $f_c = W / 3 / 7.67$ $= W (.04348)$ $f_c = \underline{\underline{12,370 \text{ PSI}}}$					
REV. NO.	REV. DATE	AUTHOR	DATE	CHK'D. BY	DATE	CHK'D. BY	DATE

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S.O.	PTNP-188	CALC. NO.	8/92	JW Richard	2/93				
		FILE NO.			GROUP	CHE			
<h3>ROTO-LOCK BACKFIT</h3> <p>THESE ITEMS SEE ONLY THE LOAD FROM THE INTERNALS. THE LOWER INTERNALS WEIGH <math>W_L = 260,000</math> LB. THE UPPER INTERNALS WEIGH <math>W_U = 120,000</math> LB. ALL PARTS EXCEPT THE INSERTS ARE USED TO LIFT BOTH THE UPPER AND LOWER INTERNALS SO <math>W = 260,000</math> LB FOR THESE ITEMS. WHEN ANALYZING THE ENGAGEMENT OF THE STUDS TO THE INSERTS IT WILL BE CONSERVATIVELY ASSUMED THAT 9 OUT OF 8 LANDS ARE ACTIVE WHEN LIFTING THE LOWERS, AND 6 OUT OF 9 LANDS WILL BE ASSUMED ACTIVE WHEN ANALYZING THE LIFT OF THE UPERS.</p>									
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PROJECT PSE/PNJ	AUTHOR <i>[Signature]</i>	DATE 2/83	CHK'D. BY <i>[Signature]</i>	DATE 2/83	CHK'D. BY <i>[Signature]</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE		
ROTO-LOCK BACKFIT LOAD NUT A (31)			<p>THREAD SHEAR</p> $f_v = F/A_v = (W/3)/A_v$ $A_v \approx D_p \ell \pi/2$ $D_p = D_s - .64952/n$ $= 4.250 - .64952/4$ $= 4.0976$ $p = 3.00$ $A_v = 4.0976(3.00)\pi/2$ $= 19.262 \text{ in}^2$ $f_v = W/3/19.262$ $= W(.017305)$ $f_v = \underline{\underline{4499 \text{ PSI}}}$		
MATERIAL: ASTM A 276 TYPE 304 HOT ROLLED COND. A					
			<p>BEARING</p> $f_c = F/A_c$ $P = W/3$ $A_c = 5.50(4.75) - 2(.75 + .167)^2$ $= \pi/4(4.250)^2$ $9.9205 \text{ in}^2$ $f_c = W/3/9.9205$ $= W(.03360)$ $f_c = \underline{\underline{8736 \text{ PSI}}}$		
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PROJECT PSE/PNJ	AUTHOR <i>[Signature]</i>	DATE 1/23/83	CHK'D. BY <i>[Signature]</i>				
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE				
ROTO-LOCK BACKFIT ADAPTOR (32) A & B		UPPER THD. SHEAR FOR PSE IT IS THE SAME AS FOR THE ROTO-LOCK BACKFIT LOADNUT $f_v = 4499 \text{ PSI (PSE)}$ FOR PNJ $f_v = P/A_v = W/3/A_v$ $A_v = D_p \pi/2$ $D_p = D - .04952/n = 4.500 - .04952/4 = 4.338$ $f_v = W/3/(4.338 + 300 \cdot \pi/2) = W(.01631)$ $f_v = 4240 \text{ PSI (PNJ)}$					
MAT'L: ASTM A 276 TYPE 304 HOT ROLLED COND A.		TENSION @ MINIMUM AREA $f_t = F/A_t = W/3/A_t$ $A_t = \pi/4 (d_o^2 - d_i^2)$ PSE $d_o = 3.897$ PNJ $d_o = 4.147$ $d_i = 1.75$ $A_{tPSE} = 9.5222 \text{ in}^2$ $A_{tPNJ} = 11.102 \text{ in}^2$ $f_{tPSE} = W/3/9.5222 = W(.0350\%)$ $f_t = 9,102 \text{ PSI (PSE)}$ $f_{tPNJ} = W/3/11.102 = W(.0300\%)$ $f_t = 7,807 \text{ PSI (PNJ)}$					
		LOWER THREAD SHEAR THE LOWER THREAD SHEAR IS THE SAME AS FOR THE GUIDE SLEEVE: $f_v = 4475 \text{ PSI}$					
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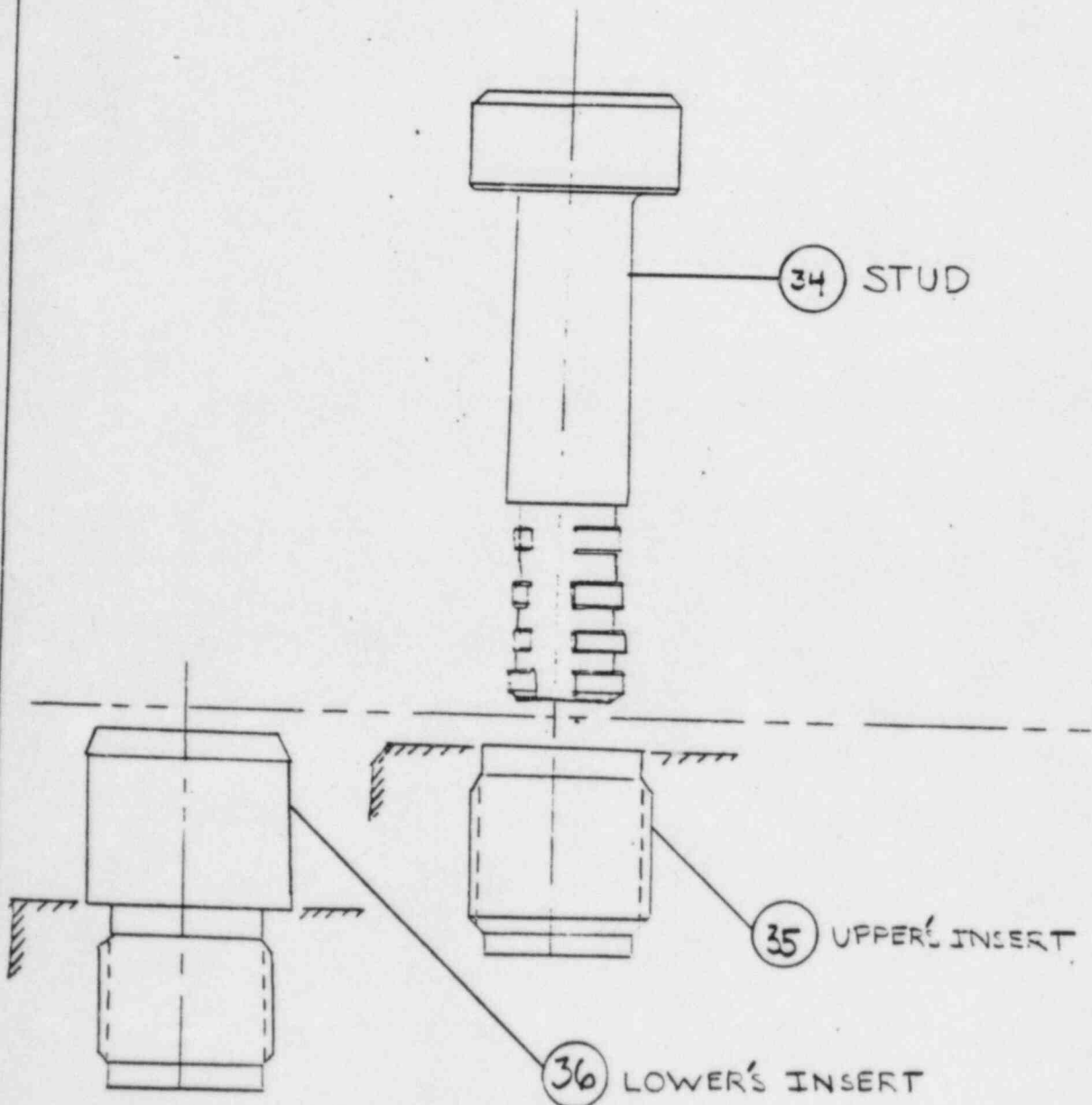
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PROJECT PSE/PNJ	AUTHOR <i>[Signature]</i>	DATE 2/7/83	CHK'D. BY <i>[Signature]</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE
<p>ROTOLOCK BACKFIT (33) GUIDE SLEEVE</p>		<p>TENSION &amp; THD RELIEF</p> $f_t = P/A_t = (W/3)/A_t$ $A_t = \pi/4 (d_o^2 - d_i^2)$ $= \pi/4 (6.50^2 - 5.504^2)$ $= 8.869 \text{ in}^2$ $f_t = W/3 / 8.869 \text{ in}^2$ $= W(.037586)$ $f_t = \underline{\underline{9,772 \text{ PSI}}}$	
<p>MAT'L: ASTM A276 TYPE 304 HOT ROLLED, ANNEALED &amp; PICKLED COND A.</p>		<p>THREAD SHEAR</p> $f_v = P/A_v = W/3/A_v$ $A_v = \pi D_p l / 2$ $D_p = D_s - .64952/n$ $= 5.5 - .64952/4$ $= 5.3376 \text{ in}$ $l = 2.75 - .44 = 2.31$ $A_v = \pi (5.3376)(2.31) / 2$ $= 19.368 \text{ in}^2$ $f_v = W/3 / 19.368 = W(.01721)$ $f_v = \underline{\underline{4475 \text{ PSI}}}$	
		<p>BEARING ON STUD</p> <p>THE BEARING ON THE STUD IS THE SAME AS THE BEARING OF THE STUD ON THE GUIDE SLEEVE.</p> $f_t = \underline{\underline{9889 \text{ PSI}}}$	
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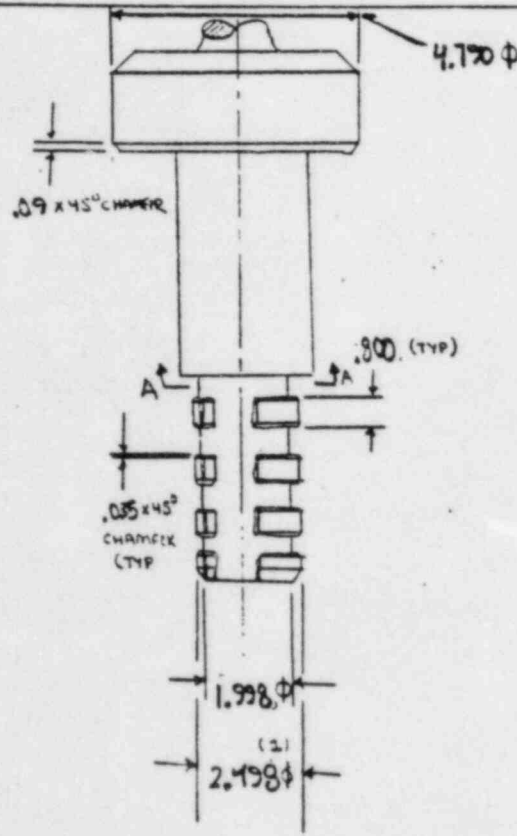
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PROJECT PSE/PNJ	AUTHOR <i>J. W. Richard</i>	DATE 2/79	CHK'D. BY <i>J. W. Richard</i>
S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE

ROTO-LOCK BACK FIT



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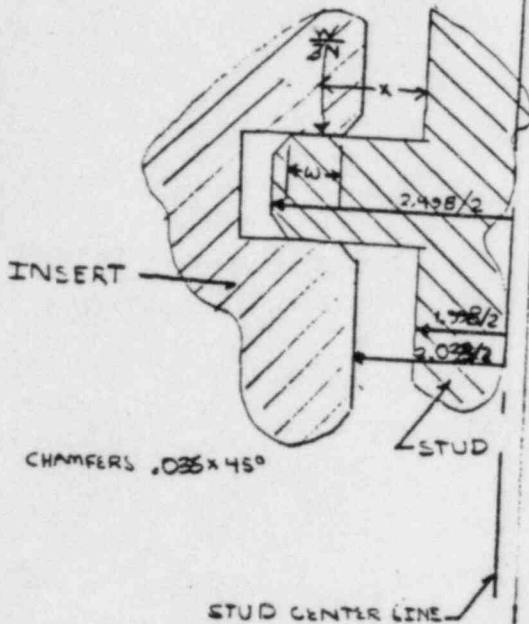
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PROJECT PSE/PNJ		AUTHOR <i>J. Richard</i>		DATE 2/83		CHK'D. BY <i>J. Richard</i>		DATE	
S.O. PTNP-188		CALC. NO.		FILE NO.		GROUP CHE			
STUD (34)				TENSION AT SEC A-A (MIN. DIA)					
MAT'L: ASTM A564, TYPE 630, 17-4 PRECIPITATION HARDENING SS. AISI TREATED @ 1100°F FOR 4 HRS & AIR COOLED. $T_{min} = 140,000$ PSI				$f_t = P/A_t$ $P = (W/3) = 260,000/3$ $A_t = \pi/4 d^2 = \pi/4 (1.998)^2 = 3.135 \text{ in}^2$ $f_t = W/3 / 3.135$ $= W / (.10632)$ $f_t = \underline{\underline{27,642 \text{ PSI}}}$					
									
				SHEAR OF THE LANDS $f_v = F/A_v = (W/3)/A_v$ $A_v = L_c d$ $L_c = \text{LENGTH OF LANDS}$ $d = \text{DEPTH OF LANDS}$ $.800 \text{ in}$ $L_c = \frac{54}{360} \pi (2.998) \times N$ $N = \text{NUMBER OF LANDS, LETTING THE LARGEST BOTTOM LAND BE EQUIVALENT TO THE TYPICAL TOP LANDS.}$ $(\frac{W}{N})_{max} = \frac{260,000}{9} = 28,889$ $L_c = 8.474 \text{ in}$ $A_v = 8.474 (.800) = 6.779 \text{ in}^2$ $f_v = (260,000/3) / 6.779$ $= W / (.04917)$ $f_v = \underline{\underline{12,784 \text{ PSI}}}$					
				BEARING ON GUIDE SLEEVE $f_c = P/A_c = (W/3)/A_c$ $W_{max} = 260,000 \text{ lb}$ $A_c = \pi/4 [(4.790 - 2(.09))]^2 - (2.937 + 2(.12))^2$ $= 8.764 \text{ in}^2$ $f_c = 260,000/3 / 8.764$ $f_c = \underline{\underline{9889 \text{ PSI}}}$					
(2) DIAMETER OF CIRCLE ENVELOPING OUTER SURFACE OF LANDS. LAND HAVE 3 TEETH PER LEVEL EQUALLY SPACED ABOUT IT. EACH TOOTH IS .54 DEPTH WIDE.									
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BENDING IN LANDS



w = width of bearing surface  
 $= 2.498/2 - .035 - [2.038/2 + .035]$   
 $= 0.160 \text{ in}$

x =  $2.498/2 - .035 - .160/2 - 1.998/2$   
 $= .135 \text{ in}$   
 = moment arm of force

$f_b = M/Z$

$Z_{\text{LAND}} = Lcd^2/6$   
 $= \frac{54}{300} \pi (1.998)(.800^4)/6$   
 $= .10043 \text{ in}^3$

M = bending moment  
 $= Px = \left(\frac{W}{3N}\right)(.135)$

N = 9  
 $M = (W/3/9) * .135 = .00500W$

$f_b = .005 W / .10043$   
 $= W(.04979)$   
 $f_b \underline{12,244 \text{ PSI}}$

BEARING ON LAND SURFACE

$A_1 = (2.498 - 2(.035))^2 \pi/4$

$A_2 = [2.033 + 2(.035)]^2 \pi/4$

$A_c = A_1 - A_2 = 1.1566 \text{ in}^2$

$A'_c = N \frac{54}{300} A_c - 2N(w + .04)$

N = 9

$A'_c = 1.446 \text{ in}^2$

$f_c = P/A'_c$

$= (W/3)/A'_c$   
 $= W(.2305)$

$f_c \underline{59,929 \text{ PSI}}$

NOTE:

The stresses in shear, bending, or land bearing are dependent on  $\frac{W}{N}$ .  $\frac{W}{N}$  is maximum when lifting the lowers, so the higher stresses occur when lifting the lowers. Therefore, for the stud stresses  
 W = 260,000 lb  
 N = 9 out of 12, assumed to live

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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE
UPPER'S INSERT <span style="border: 1px solid black; border-radius: 50%; padding: 2px;">35</span>		<p>THREAD SHEAR</p> $f_v = F/A_v = W/3/A_v$ $A_v = [D_3 - .64952/n] \pi/2$ $= (3.20 - .64952/4)(4.50 - .75 - .37) \pi/2$ $= 17.720 \text{ in}^2$ $f_v = W/3 / 17.720 = W(.01881)$ $f_v = \underline{\underline{2257 \text{ PSI}}}$	
<p>MAT'L:</p> <p>ASTM A637 GRADE 683 TYPE 2</p> <p>HUNTINGTON ALLOYS X-750 HOT FINISHED</p> <p>ROUND 115,000 PSI MINIMUM</p> <p>YIELD STRENGTH</p>		<p>BENDING IN LANDS</p> <p>FROM STUD CALLS:</p> <p><math>w = .160</math> = width of bearing surface</p> <p><math>x</math> = moment arm of force</p> $= 2.533/2 - .035 - .160/2 = 2.033/2$ $= .135 \text{ in}$ $f_b = M/Z$ $Z_{\text{LAND}} = L d^2 / 6 = \frac{54}{360} \pi (2.533)(.575)^2 / 6$ $= .065775 \text{ in}^3$ <p><math>M</math> = bending moment</p> $= (\frac{W}{3N}) (.135)$ <p><math>N = 6</math></p> $M = W(.0075)$ $f_b = .0075 W / .065775$ $= W(.11403)$ $f_b = \underline{\underline{13,683 \text{ PSI}}}$	
		<p>SHEAR OF LANDS</p> $f_v = P/A_v = W/3/A_v$ $A_v = \frac{54}{360} N \pi (2.533) = 7.1619 \text{ in}^2$ $f_v = W/3 / 7.1619 = W(.04654)$ $f_v = \underline{\underline{5585 \text{ PSI}}}$	
<p>THE BEARING STRESS IS THE SAME AS FOR THE STUD EXCEPT <math>w = 120,000 \text{ PSI}</math> AND <math>N = 6</math></p> $f_c = W(.2305) \frac{9}{8}$ $f_c = \underline{\underline{41,490 \text{ PSI}}}$			
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S.O. PTNP-188	CALC. NO.	FILE NO.	GROUP CHE				
<p>LOWER'S INSERT (36)</p> <p>MAT'L: ASTM A 637 GRADE 688 TYPE 2 HUNTINGTON ALLOYS X-750 HOT FINISHED. ROUND. 115,000 PSE MINIMUM YIELD STRENGTH</p>		<p>THREAD SHEAR</p> $f_v = P/A_v = W/3/A_v$ $A_v = (D_s - .64952/n) \pi t / 2$ $= (3.50 - .64952/4) (3.250 - .50 - .5) \pi / 2$ $= 12.478 \text{ in}^2$ $f_v = W/3/12.478 = W(.02671)$ $f_v = \underline{\underline{6946 \text{ PSI}}}$					
		<p>BENDING IN LANDS</p> <p>THE BENDING IN THE LANDS IS THE SAME AS FOR THE UPPER INSERTS EXCEPT <math>W = 260,000 \text{ LB.}</math> &amp; <math>N = 9</math></p> $f_b = W(.11403) \frac{6}{9}$ $f_b = \underline{\underline{19,765 \text{ PSI}}}$					
<p>THE BEARING STRESS IS THE SAME AS FOR THE STUD</p> $f_c = W(.2305)$ $f_c = \underline{\underline{59,929 \text{ PSI}}}$		<p>SHEAR OF LANDS</p> <p>THE SHEAR IN THE LANDS IS THE SAME AS FOR THE UPPER INSERTS EXCEPT THAT <math>W = 260,000</math> and <math>N = 9</math></p> $f_v = W(.04654) \frac{6}{9}$ $f_v = \underline{\underline{8067 \text{ PSI}}}$					
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