NUREG/CR-2350

German Standard Problem 4a



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ANCO Engineers, Incorporated

Prepared for U.S. Nuclear Regulatory Commission

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German Standard Problem 4a

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ABSTRACT

This study deals with structural pip ng response to dynamic loads generated by hybraulic transients in nuclear power plants. Transients were induced by means of a rupture disc and closure of a feedwater check valve at the Heissdampfreaktor, West Germany. Blind predictions of piping response were made using the computer code EASE2. Comparisons between computer simulations and experimental observations are contained in the report.

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PREFACE

The following study was conducted by ANCO Engineers, Inc. for the Division of Reactor Safety Research (Dr. John O'Brien, Project Officer), Nuclear Regulatory Commission as part of a joint research effort supported by the NRC and the Ministry of the Interior (BMI) of the Federal Republic of Germany. This report is one of a series of studies conducted by ANCO Engineers as part of the Heissdampfreaktor safety research program. Primary responsibility at ANCO for the work reported herein rests with Mr. William B. Walton, Dr. George H. Howard, and Mr. Blake Johnson. The authors gratefully acknowledge the support provided by Dr. W. Winkler and Dr. T. Grillenberger of Gesellschaft fur Reaktorsiecherheit, Garshing, as well as Dr. L. Malcher and Dr. Katzenmeier of Kernforschungszentrum Karlsruhe.

1.0 INTRODUCTION

At the request of the Ministry of the Interior (BMI) Federal Republic of Germany, the Society for Reactor Safety (GRS) formulated suggestions dealing with, among other things, the formulation and execution of a Standard Problem in the area of pressure wave propagation in nuclear power plant piping systems. Standard Problem No. 4 (DSP4) was defined and funded. The aim of this problem was to study the closing behavior of a feedwater check valve and the fluid dynamics in its respective pipeline as a result of a simualted break in the pipeline and ensuing rapid closure of the check valve. The tests were to be performed at the Heissdampfreaktor (HDR) using the URL pipe system.

In connection with this study an additional German Standard Problem (4a) was defined. This study was to be a part of the same tests performed for problem 4. The emphasis here would be the determination of the structural dynamic behavior of the pipe system (measurement and calcualtion of pipe displacements, accelerations, strains/stresses) when subjected to fluid forces from the problem 4 event.

As a part of German Standard Problem 4a (DSP4a), ten (10) engineering firms performed "blind" calculations to predict the structural dynamic response of the URL piping system to problem 4 fluid forces. The results of these calculations were sent via magnetic tape to GRS prior to distribution by GRS of the embargoed experimental results. Chapter 6 of this report presents a very brief compt son of ANCO's predictions to the subsequently released data. ANCO Engineers, Inc. was one of those firms and the sole american participant. This report documents the tarks performed in predicting the URL response.

2.0 DESCRIPTION OF GERMAN STANDARD PROBLEM 4a (DSP4a)

DSP4 was an experimental and theoretical simulation of an assumed break in a feedwater line for a nuclear power plant. The simulation was performed using a leg (portion) of the modified recirculating loop of the Heissdampfreaktor (HDR) at Kahl, West Germany (see Figure 2.1). The pipe system was modified by rebuilding a portion of the external forced circulation loop. This involved installing circulation pump Z 199 (Item 7 in Figure 2.1).

The test involved circulating compressed water* through a portion of the pipe system. The circulation started from the S-support 3 (see Figure 2.1) and travelled to a spherical piece 12, and on through locations 6, 7, 5, 11, 10, 4, 9, and back to the reactor vessel 2. Once steady-state fluid conditions were achieved, the break in the feedwater line was simulated. This involved: (1) circulation pump shut off; (2) fracture of the rupture disc (8 in Figure 2.1, simulated assumed break); (3) closing of rapid shutoff valve 5; and (4) eventual closure of the check-valve 4. At check-valve closure, pressure waves were generated which propogated through the system.

The pressure waves generated during the test were of such amplitude that a high stress state in the pipe material resulted (high relative to the yield stress of the material). For this reason it was of importance to understand the structural dynamic behavior of the pipe system during the fluid dynamic event. This is where DSP44 ties in with DSP4. For DSP44, the structural dynamic behavior of the URL test pipe line was to be theoretically determined; applied structural loading was to be determined from the

^{*}Note: The state of the water in the test line during the circulation phase was 70 bar (1015 psi), 220°C (428°F), and 4 m/sec (13.1 ft/sec; in opposite direction to blowdow. flow direction).

FIGURE 2.1: MODIFIED PRIMARY COOLANT LOOP AT HDR



DSP4 test data.* During the blowdown test the URL pipe was instrumented with displacement transducers, accelerometers, and strain gauges. The data from these instruments is the basis against which all DSP4a theoretical simulation results are to be compared.

The portion of the URL system involved in the structural simulation is shown in Figure 2.2. The pipe has a total length of 18.80 meters (6.6) ft). The pipe begins at the reactor vessel and travels to a rupture disk at its other end. There is a fixed point near the rupture disk. There are no structural supports (snubbers or struts) attached to the pipe. The dimensions and material specification of the various sections of the pipe are given in Table 2.1. The inner pipe diameter and wall thickness vary from 0.351 to 0.453 meter (1.152 to 1.486 ft) and 0.014 to 0.142 meter (0.046 to 0.466 ft), respectively.

As a part of the theoretical simulation for DSP4a, the specified response quantities are to be calculated; they are displacement, acceleration, and stress. The components of stress of interest are the bending, tensile, torsional, tangential, and comparative (von Mises) stresses. Also, the bending axis angle was desired. Figure 2.3 and Table 2.2 describe the desired quantities.

1

The fluid forces on the pipe, due to the pressure waves, were to be calculated using the DSP4 test data (i.e., fluid pressure). The pressure as a function of time is known at numerous locations, particularly at inlets and/or outlets of pipe elbows. The method used for obtaining the forces was to be developed by each investigator. Thus, DSP4a involves determining both the fluid/structure interface forces and the structural response.

2-3

^{*}Wote: The calculations performed by ANEO Engineers. Inc. were to simulate reality as closely as possible (they were not designed to be conservative). Also, no previous (prior to completion of the analysis) knowledge of the structure dynamic test results was known.



FIGURE 2.2: Portion of URL System Used for Structural Dynamic Simulation.

TABLE 2.1

DIMENSIONS	AND	MATERIAL	DESIGNATIONS	FOR	PIPE	SECTIONS
the other substances and the second substances		a second second second second second				

Lfa.* Nr.	Teilstück	Werkstoff	Nennweite	Innendurch- messer [m]	mittlere x) Wanddicke ['mm]	Teilstück- länge [mm]
1	RDB-Wand	23NiMoCr36	-	-	142	142
2	T-Stutzen	23NiMoCr36	NW 350	351,2	104,4/27,6	375
3	120°-Boyen	Nr. 4550	Ni 350	360	25,0	1120
4	Rohrstück mit 30°-Biegung	Nr. 4550	NW 350	360	19,3	2508
5	60°-Bogen	Nr. 4550	NW 350	360	25,0	560
6	Rohrstück	Nr. 4550	NW 350	360	19,3	1494
7	90°-Bogen	Nr. 4550	NW 350	360	25,0	840
8	Rohrstück	Nr. 4550	NW 350	360	19,3	630
9	Probenstück	WB 35	IW 350	377	14,0	1000
10	Rohrstück	Nr. 4550	NW 350	360	19,3	2925
11	Übergangsstück	15Mo3	NW 350	360	19,3/25,0	100
12	90°-Bogen	15Mo3	NW 350	356,4	25,0	957
13	Rohrstück	15Mo3	NW 350	371,4	17,5	510
14	MeBring I	15Mo3	-	371,4		480
15	SRV 350	GS C25	NW 350	-	-	1250
16	Meßring II	15Mo3	-	371,4	-	500
17	Rohrstück	15Mo3	IW 350	371,4	17,5	260
13	61°-Bogen	15Mo3	NW 350	356,4	25,0	648
19	T-Stück	15Mo3	17/450/350	428/371,4	40,0/17,5	1300
20	Stutzen	15Mo3	NW 450	453	61,0	520
21	Meßring III m. Bruchstutzen	15Mo3	NW 450	453	27,5/93,5	740
22	Berstscheiben u. Mündungsrohr		NW 450	453	-	250
						19109

x) Die mittlere Panddicke wurde aus den Angaben der Rohrleitungszeichnungen entnommen und weicht erheblich von den an einzelnen Stellen durchgeführten MPA-Messungen ab (siehe Anlage 12).

Die beim Probenstück (1fd. Nr. 9) angegebenen Maße sind Istmaße.

* Note: These sections are defined in the next page of this table.



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²⁻⁶



FIGURE 2.3: Desired Structural Response Quantities.

2.7

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TABLE 2.2: DESIRED STRUCTURAL RESPONSE QUANTITIES

Entsprechende Meßstelle	zu ermittelnde Größe		
E 2 2201	Weg in v-Richtung	~	
De Stat	Weg in v-Richtung	1	(3.4 m vom
RD CEUE	Weg in T-Richtung	5	POR-Stutzen)
R5 2203	weg in z-Richtung		NOD DIDIZENI)
RS 2204	Weg in y-Richtung)	14 4
RS 2205	Weg 1 z-Richtung	1	(1,7 m vom
RS 2206	Weg in x Richtung	7	RDB-Stutzen)
55 4004	Wed in z-Richtung	~	
SS 4005	Weg in x-Richtung	1	am Ventil
55 4005 66 4006	Wee in v-Richtung	5	
	Reg III y Richtung	1.	
55 4001	seschleunigung in x-Richtung	L	am Vantil
5S 4002	Seschleunigung in y-Richtung	1	am venu
SS 4003	Beschleunigung in z-Richtung	1	
SK 1010	Zugspannung	1	
RK 2010	Biegespannung	1	
DK 2210	Winkel der Biegeachse	11	
RK 2010	Torrionsconnouna	1	MeRebene A
RK JUIU	Torsionsspannung	1	HERENEINE A
RK 4110	langentialspannung ourch	1	
	Innendruck		
RK 5010	Vergleichsspannung	Ĩ	
RK 1011	Zuospannung	1	
DV 2011	Biegespanoung	1	
RK 2011	Wiekel der Biegeschre	1	
KK ZZII	WINKEL DEL Diegeacrise	1	MaRahana C1
RK 3011	Torsionsspannung	1	Medebelle CI
RK 4111	Tangentialspannung durch	1	
	Innendruck		
RK 5011	Vergleichsspannung	1	
RK 1012	Zuospannung	1	
DK 2012	Biegespannung	1	
RR 2012	Winkel der Biegesche	1	
RK 2212	Tanaiana and biegeactive	1	MaRahana M1
RK 3012	Torsionsspannung	1	menenene mit
RK 4112	langentialspannung durch	1	
	Innendruck		
RK 5012	Vergleichsspannung	ć.,	
RK 1013	Zugspannung	2	
PK 2013	Biegespannung	1	
DV 2213	Winkel der Riegeachse	1	
KN 2213	Tarrianschapping	1	Melehene F
RK 3013	Torsionsspannung	1	medebene E
RK 4113	langentialspannung our en	1.	
	Innendruck	1	
RK 5013	Vergleichsspannung	ć.,	
PK 1014	Zuospannung	7	
DV 2014	Riegespannung	1	
RN 2014	Wield der Biegeschen	1	
RK 2214	WINKE DER Diegeachse	1	MaRabana E
RK 3014	Torsiensspannung	1	Mebebene P
RK 4114	Tangentialspannung durch	1	
	Innendruck	1	
RK 5014	Vergleichsspannung	1	

3.0 STRUCTURAL MODEL OF THE URL PIPE SYSTEM

To simulate the dynamic behavior of the URL pipe structure for DSP4a, a linear finite element model was constructed. A plot representing the model is shown in Figure 3.1. The model includes the section of the pipe from the reactor vessel wall to the T-piece. Fixed points (all degrees of freedom deleted) were assumed at each end of the pipe. The model was defined using 30 nodes (end nodes fixed; 163 degrees of freedom) and 27 finite elements (straight and curved pipe elements). Appendix A contains a partial listing of the EASE2 input data deck which defines the finite element model.

Various coordinate systems were defined. The global coordinate system (the reference system for mass and st. fness matrix assembly) is given by X, Y, Z (see Figure 3.1). Local coordinate systems were defined for output of displacement and acceleration simulation results in accordance with DSP4a definition (specified by the Society for Reactor Safety; T. Grillenberger); they are shown in Figure 3.2 and are represented by x, y, z.

A flexibility factor k_p is used in EASE2 to modify the bending terms in the flexibility matrix for the curved pipe element. The flexibility matrix for the curved pipe element is derived totally from curved beam theory. This overestimates the stiffness (as compared to experimental data). To correct for this, the flexibility factor k_p is used. It is given by:

$$K_{\rm p} = (1.65/h)/[1 + (6p/Eh)(R/t)^{4/3}] \ge 1$$

^{*}Note: The computer code EASE2 (Elastic Analysis for Structural Engineering) was used. This code is accessible through the CDC Cybernet Network Worldwide.





where: p = internal pressure h = tR/r² r = (d₀-t)/2 R = bend radius t = wall thickness d₀ = outside pipe diameter

E = Young's modulus

This correction can be significant. From experience, errors on the order of 25% have been observed in the first natural frequency of piping systems when this correction was not included.

Following model formulation, an eigenvalue analysis of the URL model was executed. The first ten (10) eigenfrequencies are given in Table 3.1. It is likely that with the number of nodes used in defining the pipe model (30 nodes) that the last few modes are not accurately defined. If it is minimally acceptable to use four points to define a half cycle of a standing wave, then the seventh (7th) mode is approximately the highest mode that can be acceptably defined with the thirty (30) node points. Four points will reasonably well define a half cycle. For three points, it is possible to define up to about the tenth (10th) mode. Three points will marginally define a half cycle. For these reasons, it is reasonable to expect the URL modes to be fairly well defined up to at least the eight (8th) or ninth (9th) mode.

1

TABLE 3.1

NATURAL FREQUENCIES FOR HDR/URL BLOWDOWN MODEL

Mode #	Eigenfrequency (Hz		
1	5.46		
2	8.54		
5	9.43		
	26.18		
5	30.55		
6	39.71		
7	45.57		
8	50.54		
9	64.23		
10	76.22		

4.0 FRESSURE WAVE FORCES ON THE URL PIPE

DSP4a involves accurately simulating the structural dynamic response of the modified URL system when excited by the DSP4 blowdown fluid forces. This type of simulation deals with fluid-structure interaction (the response of a structure due to transient fluid events).

There are two general approaches to the fluid-structure problem. One involves the difinition and solution of a coupled problem. For this approach, the equations representing the fluid and structure behavior are coupled to each other. In order to solve for the response of either the fluid or structure, the response of both must be found simultaneously. The other approach involves determining an uncoupled solution. This involves solving for or determining experimentally the fluid response, independently of the structural response. The fluid response is then input into the structural problem, and the uncoupled structural response is determined. The latter of these two approaches was used for DSP4a. This is because the fluid response, determined as a part of DSP4, is to be used as input to DSP4a. The fluid response (i.e., pressure) is to be used to determine the fluid forces on the URL pipe. These forces will be used in determining the structural response.

Given the transient lluid response of a system, there are at least two possible basic approaches for determining the fluid forces (uncoupled fluidstructural problem) on a pipe system. One approach could be called the "equivalent concentrated forces method." Basically it says, given a continuous fluid stress distribution in a pipe (at the pipe fluid interface), it is possible to determine equivalent concentrated loads at the node points used to define the structural model. This is done, in part, by applying the following equation:

$$P_{eq} = \int a^T \phi dS$$

where P_{eq} is the equivalent end load vector for a pipe element, a^{T} is the transpose of the matrix a, where a gives the relationship, for a discrete structural element (i.e., finite element), between the spatially continuous interior displacements and the discrete element displacements fat the node points), ϕ is the matrix of distributed surface stresses (distributed field of fluid stress), and dS refers to the integration being carried out over the applicable surface (s). This expression is only for a single element. A P_{eq} would have to be determined for each element. Then, a global P_{eq} would need to be determined for the entire structure. This is an excellent approach, provided the distributed loading can be determined for the entire structure. This may be possible for the DSP4a, but would be very difficult, if at all possible. As this is a wave propagation problem, it greatly complicates the determination of a continuous pressure distribution. Because of this, it was decided to use a second approach to solving the uncoupled problem.

Another of the basic uncoupled fluid-structure approaches involves the use of the linear momentum equation from fluid mechanics (control volume formulation):

 $\overline{F}_{S} + \iiint \overline{B}(\rho dv) = \iint \overline{V}(\rho \overline{V} \cdot d\overline{A}) + \frac{\partial}{\partial t} \iiint \rho dv$ C.V. C.S. C.V.

where \overline{F}_s is the total surface force on the control surface (C.S.), \overline{B} is the body force distribution, \overline{V} is the fluid velocity, $d\overline{A}$ is a differential area normal to the C.S. and ρ is the fluid density.

To apply this equation, the space occupied by the fluid in the pipe was subdivided into control volumes (the union of the control volumes is equal to the total space in the pipe). There are an infinite number of ways to subdivide the space. Various methods for doing this have been devised.

The control volumes for this problem obviously had to be contained in straight, curved, and combinations of straight and curved lengths of pipe (see Figure 4.1). One constraint on the selection of volumes was that the pressure had to be known at each of its two ends. (As will be discussed later, this was not necessary for straight volumes.) This is necessary for evaluating the surface force \overline{F}_{s} .

The question arises as to how large the control volumes can be (or to how small they should be). For each control volume, there is essentially one resultant force. If there are only a few control volumes, there will only be a few forces representing the spatially continuous load distribution on the pipe. The smaller the control volumes the more closely their combined loading resembles the continuous loading.

In choosing the size of a control volume it is important to consider the degree to which the center of pressure of its respective fluid changes location during the fluid dynamic event. The control volume resultant force should be applied at the center of pressure. This can be done by having a fine mesh for the discrete structural model, especially for such elements as curved pipes (locations where some of the largest resultant forces are generated). During the structural simulation the resultant force can be moved from node to node. Even though these calculations can be performed with enough data, they are very involved (i.e., the center of pressure must be calculated at each solution step). This, coupled together with the fact that only a relatively limited amount of DSP4 data is available, the resultant control volume force, for DSP4a, was not applied at the center of pressure, but approximately at the geometric center of the control volume. If the control volumes are chosen to be ico large there will possibly be considerable error in the force application point locations.



4-4

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24

5.8

On the basis of the above discussion, the URL blowdown model (DSP4a) was broken up into six (6) control volumes (see Figure 4.2). As can be seen, four of the volumes are made up of a combination of straight and curved sections. Two of the volumes are straight. The point of application of the resultant force for a given control volume is represented by the designator LP (load point). No load point is defined for volume six (6). This volume is straight with the only forces it can exact on its section of pipe being frictional. For a DSP4 type of event, the frictional fluid forces are negligible.

In determining the control volume forces on the pipe, it is necessary to evaluate the momentum efflux rate, rate of change of control volume momentum, and body force integrals for each control volume. The two momentum integrals are dependent on mass flow rates. The major events that take place during the DSP4 test that might effect the mass flow rate are: (1) the circulation pump gives a constant flow rate of compressed liquid of approximately 1600 m³/hour (15.7 ft /sec); (2) the pump is shut off at the beginning of the test, resulting in essentially zero flow rate; (3) fracture of the rupture disc, resulting in flashing of the compressed liquid at the disc; (4) closing of the rapid shut-off valve immediately after "uputure disc fracture (this insures isolation of the second section of pipe); and (5) closing of the check-valve SRV350, resulting in zero mass flow rate between the fitting and reactor vessel (after closure). The flashing will give rise to large steam and small to moderate compressed liquid velocities. This will generate some force on the pipe, but is not the major forcing generated during the entire event. The major forcing is generated during check-valve closure. For this phase of the forcing, the compressed liquid flow rate is initially at a small to moderate level. As the event progresses, the flow rate will diminish, eventually, to zero. Figure 4.3 gives a hypothetical example of this. With the portion of the DSP4 event that gives rise to the largest forcing, seeing only from moderate to zero net flow rates, a reasonable assumption for calculating peak forces on the URL pipe may be that the fluid flow terms in the momentum equation can be neglected. Also, it is reasonable to assume that the body force distribution will not materially affect the flow.



FIGURE 4.2. Control Volumes Used for DSP4a.



FIGURE 4.3: Hypothetical Example of A DSP4 Event.

Using these assumptions, the momentum equation (control volume formulation) reduces to $\overline{F}_{s} = 0$ (the sum of the external surface forces equals zero). This equation is applied to an arbitrarily shaped control volume (for a piping system) as shown in Figure 4.4. The only forces of concern are the pressure forces (forces on the control volume from the adjacent fluid) and the force of the pipe on the volume. It is seen that the resultant force of a control volume (using the above assumptions) on its corresponding section of pipe is given by:

$$\overline{F}_{R} = p_{1}A_{1}\hat{1}_{1} - p_{2}A_{2}\hat{1}_{2}$$

This equation is applied to each of the control volumes used to define the URL space. Figure 4.5 defines the pressures and area used in the computations. The equations for the fluid forces (on the pipe) at the load points are given in Table 4.1. Appendix B gives the derivation of these simple equations.

The fluid pressure time history data was used together with the equations in Table 4.1 to calculate the fluid forces on the pipe. The channels of pressure data that were taken to correspond to P_1 ,..., P_7 are listed in Table 4.2. Plots of the calculated fluid forces on the pipe are given in Figures 4.6 through 4.15. Table 4.3 gives the force application point and direction for a given force number (i.e., force number RF0006). Table 4.4 gives the minimum and maximum values of each of the force components.



FIGURE 4.4: Arbitrarily Shaped Control Volume for Obtaining Pipe Segment Forces.



FIGURE 4.5: Pressure and Area Definitions and Fluid Force Application Locations

P_i, A_i - pressure and area i, respectively
LP_i = load point j (point of application of fluid force)

TABLE 4.1

DSP4 FLUID FORCES ON PIPE

Load Point	Fluid Force on Pipe at Load Point*
1	$\overline{F}_{1} = (P_{2}A_{2} - P_{1}A_{1}\sin 29^{\circ})\hat{1} + P_{1}A_{1}\cos 29^{\circ}\hat{j}$
2	$\overline{F}_{2} = (P_{3}A_{3} - P_{2}A_{2})i$
3	$\overline{F}_3 = P_5 A_5 \hat{i} + P_4 A_4 \hat{j}$
4	$\overline{F}_4 = P_7 A_7 \cos 60^\circ \hat{i} + P_7 A_7 \sin 60^\circ \hat{j} + P_6 A_6 \hat{k}$
5	$\overline{F}_5 = -P_7 A_7 \cos 15^\circ \hat{1} - P_7 A_7 \sin 15^\circ \hat{j}$

*Note: The unit vectors i, j, k correspond to the local coordinate systems at the load points (see Figure 4.5).

Pressure	Corresponding Transducer	Flow <u>Area</u> (m ²)
P1	RP2114	0.1083
P2	RP2108	0.1083
P3	RP2205	0.1083
P4	RP2205	0.1083
P ₅	RP2203	0.1018
P ₆	RP2202	0.1018
P ₇	RP2201	0.1018

TABLE 4.2: RELATION OF PRESSURE P: TO TRANSDUCER CHANNEL

13

Res

\$

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113

100

45

1



4-13



CHANNEL 2 RE0002 N

4-14


CHANNEL 4 RE0004 N





CHANNEL 6 RE0006 N











CHANNEL 19 RE0013 N

FIGURE 4.15

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8

51 B

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. inf .

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(Problem with file information for RF0014)

TABLE 4.3

Force Number*	Load Point (LP)	Location of Force (ANCO Node)	Direction of Force (Local Coordinates)**
RF0001	5	11	x
RF0002	5	11	у
RF0003	5	11	Z
RF0004	4	18	х
RF0005	4	18	у
RF0006	4	18	Z
RF0007	3	37	х
RF0008	3	37	у
RF0009	3	37	z
RF0010	2	*43	x
RF0011	2	43	У
RF0012	2	43	z
RF0013	1	49	x
RF0014	1	49	У
RF0015	1	49	Z

LOCATION AND DIRECTION OF APPLIED FLUID FORCES

*This corresponds to the force headers of file #2.

** See Figure 4.16 for the definition of these directions.



 $LP_j = load point j$

FIGURE 4.16: Local Coordinate Directions for Applied Fluid Forces.

TABLE 4.4

		11.4 2 2.5	an 20 1/2	194	*** ***	25	171737	1. 19 8 1 1 1 1
H X I	12 5- MIE-	VALUE	- ~ I	3.34	141.111	11	141.11	21.1-5
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Force Number	ANCO Node	Force Direction	Minimum/Maximum Value/10 ³ (N)
RF0001	11	x	-292/460
RF0002	11	У	-78/123
RF0004	18	x	-238/151
RF0005	18	У	-412/262
RF0006	18	1	-524/400
RF0007	37	х	-531/452
RF0008	37	У	-567/475
RF0010	43	x	-131/970
RF0013	49	х	-290/1550

5.0 STRUCTURAL DYNAMIC SIMULATION OF URL BLOWDOWN EVENT (DSP4a)

The finite element model discussed in Section 3.0 and the fluid forces presented in Section 4.0 were the basis for the response calculations required for German Standard Problem 4a (DSP4a). In performing the response calculations, several items were of concern; they were: (1) damping to be used; (2) the integration interval; and (3) nonlinear versus linear simulation methods.

No data was available on the damping of the portion of the URL pipe system involved in the DSP4a simulation. Damping data was available for the remainder of the pipe system. It typically varied from 3.0 to 6.0 percent of critical. It would be expected that the damping for the portion of the URL system not involved in DSP4a would be higher than that for the pipe leg of interest. This is because the leg of interest did not have any supports (sway braces, spring hangers, etc.) connected to it, whereas, the remainder of the system did. (The pipe supports generally increased the losses in the system.) For this reason it was decided to use damping values intermediate to the extreme values of 3.0% and 6.0%. The damping was chosen to be between 3.0% and 4.5% of critical. It should be noted that the transient response solution for DSP4a is not extremely sensitive to damping changes over a small damping domain (i.e., 3.0% to ^h.0%). For this reasc:, choosing intermediate damping values seems to be reasonable.

The structural equations of motion were integrated using a direct integration scheme (Newmark method). The integration interval was chosen to be equal to the discretization interval used for digitizing the DSP4 data ($\Delta t = 0.0002$ second). With this time interval and using 16 time points per cucle, it is possible to define a transient signal of up to 312.5 Hz. The fluid pressure, for DSP4 type events, generally has its major frequency content in the 50 Hz to 100 Hz range. Hence, the chosen integration

interval should be more than adequate (Nyquist sampling theorum says that a Δt of 0.0002 seconds is sufficient to detect transients up to 2500 Hz while experience indicates that such a Δt will certainly be sufficient for transients up to 1200 Hz).

The proportional damping coefficients α and β (C = α M + β K, where C, M, and K are the damping, mass, and stiffness matrices, respectively) were chosen to be 1.20 and 6.35 x 10⁻⁵, respectively. This gives an equivalent modal damping of 4.5% at 5 Hz and 100 Hz and a minimum damping of 3.0% between 5 Hz and 100 Hz.

In simulating the structural dynamic event, a linear analysis approach was selected. This decision was made because it was not known if the pipe would experience nonlinear deformation; hence, a standard approach dictated that a linear analysis be performed first, and, only if necessary, a nonlinear analysis would be performed (this would not be done for this present task).

The linear analysis was performed and the plotted results shown in Figures 5.1 to 5.42. Table 5.1 gives the minimum and maximum values for the structural response quantities specified by DSP4a. Appendix C lists the computer code that was written to process the EASE2 output. It was used to compute additional stress information and perform the necessary coordinate transformations of the displacement and acceleration results.

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CHANNEL S RS2205 MM





CHANNEL 7 SS4004 MM



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FIGURE 5.14

(Problem with File Information for RK 2010)







FIGURE 5.17

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(RK 4110 not computed)

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FIGURE 5.18

(RK 5010 not computed)



FIGURE 5.20

(Problem with File Information for RK 2011)





FIGURE 5.23 (RK 4111 not computed)
FIGURE 5.24

(RK 5011 not computed)



FIGURE 5.26

(Problem with file information for RK 2012)













FIGURE 5.32

(Problem with file information for RK 2013)











CHANNEL 16 RK1014 N/MM2

FIGURE 5.38

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(Problem with file information for RK 2014)





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TABLE 5.1

EXTREME VALUES OF STRUCTURAL RESPONSE DISPLACEMENTS AND ACCELERATIONS

Response Number	ANCO Node	Response Direction	Minisem/N Valu	Maximum 1e
RS2201	14	у	-49/44	(mm)
RS2202	14	x	-20/20	(mm)
RS2203	14	2	-47/43	(mm)
RS2204	22	у	-87/67	(mm)
RS2205	22	z	-39/43	(mm)
RS2206	: 22	х	-50/41	(mm)
SS4004	43	Z	-18/21	(mm)
SS4005	43	х	-1/2	(mm)
SS4006	43	У	-15/18	(mm)
SS4001	42	х	-38/7	(g's)
SS4002	42	у	23/26	(g's)
554903	42	Z	-19/18	(g's)

6.0 COMPARISON OF STRUCTURAL RESPONSE TEST DATA AND ANCO SIMULATION RESULTS

A brief comparison is made, herein, between some of the test data and the results of the simulation performed by ANCO Engineers, Inc. The quantities that have been compared are: (1) eigenvalues of the pipe system and; (2) maximum and minimum displacements of the pipe during the dynamic event. The test data was taken from the report "Ergebnisbericht, Blowndown-Versuch NR. V60.4.1, am 5.12.80, Versuchsgruppe SRV 350," Februar 1981, Kernforschungszentrum Karlsruhe, Projeckt HDR.

The experimentally determined and predicted eigenfrequencies are given in Table 6.1. There was a problem in comparing the two sets of values. Only two experimental frequencies were given below 25 Hz, whereas, there were three predicted frequencies. This difference was partially resolved through the following discussion. There is a large relative difference between the third and fourth theoretical eigenfrequencies. There is also a large difference between the second and fourth experimental eigenfrequencies (as defined in Table 6.1). Because of this, together with the fact that the values of the frequencies (experimental and predicted), as given for the fourth and fifth modes, are close to each other, it is believed that the experimental frequencies given in Table 6.1 for the fourth and fifth modes probably correspond to the fourth and fifth theoretical requencies.

In comparing the eigenfrequencies below 25 Hz, there are essentially two possibilities. First, that there exist only two experimental frequencies below 25 Hz; or second, that there exist more experimental frequencies below 25 Hz, one of which was not detected during data analysis. Regardless of what the situation is, it is most logical that the first experimental frequency corresponds to the first theoretical frequency. If it did not, the first experimental frequency would correspond to the second theoretical frequency giving the theoretical frequency a relative error of 73 percent. With good agreement between theory and experiment for the fourth and fifth eigenfrequencies, and with the physical system being

TABLE 6.1 COMPARISON OF EXPERIMENTAL AND THEORETICAL EIGENFREQUENCIES

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Mode	Experimental Eigenfrequency (Hz)	Predicted Eigenfrequency (Hz)	Relative Difference (%)
1	4.95	5.46	10.3
2	7.75	8.54	10.2
3		9.43	*
4	25.50	26.18	2.7
5	29.30	30.55	4.3

*Note: On the assumption that the first two experimental eigenfrequencies correspond to the first two theoretical eigenfrequencies, either there is not an experimental mode that corresponds to the third theoretical mode (9.43 Hz eigenfrequency mode) or the third experimental mode was not observed during analysis of the experimental data by German investigators. essentially linear for non-plastic deformation and hence being very amenable to modeling as a linear system, an error of 73 percent for the second frequency is improbable. Overall, there seems to be excellent agreement between the two sets of eigenfrequencies.

The maximum and minimum displacements the pipe experienced during the dynamic test (DSP4) are compared to the predicted results in Table 6.2. There is a substantial difference between experiment and theory. Some of the large difference occurs where the displacement is small (i.e., SS4005) and, hence, does not have a great deal of meaning. Some of the large difference occurs where the displacement is large (i.e., RS2204). This difference is of concern. Some of the small to moderate differences occur where the displacements range from being small to large. Overall, the comparison of theory to experiment seems to be fair, with the predicted values bounding the experimental values.

TABLE 6.2 COMPARISON OF MAXIMUM AND MINIMUM DISPLACEMENTS

	Minimum/Max	imum Value (mm)	
Transducer	Experimental	Predicted	Relative Difference (%)
RSS2202	-7/9	-20/20	186/122
RS2201	-35/37	-49/44	40/19
RS2203	-22/18	-47/43	114/139
RS2206	-/-	-50/41	-/-
RS2204	-52/62	-87/67	67/8
RS2205	(-30/30)*	-39/43	30/43
SS4005	-7/8	-1/2	85/75
SS4004	-23/27	-18/21	22/22
SS4006	-13/17	-15/18	15/6

*Mote: Displacement transducer may have been damaged during the test.

7.0 COMMENTS

Even though the DSP4a theoretical simulation was to have been as close to reality as possible it was necessary to make certain simplifying assumptions due to reasonable resource constraints and limited fluid dynamic response data from DSP4. Following is a list of these assumptions:

Structural Assumptions

- A linear structural simulation would give meaningful results.
- Each end of the pipe line was fixed.
- The pipe could be represented with pipe elements (essentially straight and curved beam elements); no ovaling of the pipe elbows would occur to any significant degree.
- The damping effects could be represented with proportional damping and the damping was between 3.0% and 4.5% of critical.
- The fluid-structure problem could be decoupled and still give reasonably correct results.

Fluid Assumptions

- The discretization of the space inside the URL pipe into the six defined control volumes (Figure 4.2) was fine enough to generate a satisfactory load distribution on the pipe.
- The net rate of momentum efflux from a control volume and the rate of change of momentum within a control volume can be neglected as compared to the fluid pressure surface force of the volume.
- Body forces are negligible.
- Choosing the geometric center of a control volume as the center of pressure will not seriously affect the results.
- No significant concentrated moments on the pipe would be generated by the control volumes.

The structural dynamic simulation was performed using the above assumptions. The response information required by DSP4a was generated. As far as can be determined by inspection of the simulation results, it appears that there are no serious problems with the solution approach or the solution.

APPENDIX A

URL BLOWDOWN FINITE ELEMENT MODEL (EASE2 INPUT LISTING)

APPENDIX A

URL BLOWDOWN FINITE ELEMENT MODEL

(EASE2 Input Listing)

Following is a listing of the data which defines the EASE2 model of the URL blowdown pipe system.

EASE2 INPUT

MASTER CONTROL PARAMETERS

HOR / URL BLOWDOWN MODEL

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PAGE 1 24 MAR 01

MASTER ECHO PRINT CONTROL EQ. *, SUPPRESS ECHO PRINT IN ALL DATA SECTIONS	1	¢	,	595	(00	15)
SOLUTION MODE CONTROL EQ. 1, STATIC SOLUTION EQ. 2, EIGENVALUE SOLUTION ONLY EQ. 3, TIME-HISTORY ANALYSIS USING MODE SUPERPOSITION EQ. 4, TIME HISTORY ANALYSIS USING DIRECT INTEGRATION EQ. 5, RESPONSE SPECTRUM ANALYSIS		ſ	4)	IMODE	(00	15-20)
STIFFNESS MATRIX RESTART CONTROL EQ. 1, PROGRAM EXPECTS TO READ THE DECOMPOSED STIFFNESS MATRIX FRGM DISK FILE TAPE12		ţ	0)	IRSTRI	(00	241
EIGENVALUES/EIGENVECTORS RESTART CONTROL EG. 1, PROGRAM EXPECTS TO READ EIGENVALUES AND EIGENVECTORS FRUM DISK FILE TAPEIO	ľ	¢.	0)	IRSTR2	t C C	251
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MAXIMUM CORE FOR EASE2 EXECUTION SLANK, DEFAULT SET TO 250000 LT. 1000008, RESET TO 1000008		(25	00001	ICORE	(00	35-401

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CCOSVC 5	CEDBAL X DS CALINDSI DE CASIEZI	CCOBVC X	2 778079 1077 1241 70 (Xa)	6-384- X 36 CV-1438 36 CV51438	SECTORAL X	Z TVEOTO NISIZO BH	A PIES OF T	000396 X 000501	2421 242164	3 78 KON 2015 X 5 3 19 K 10 K000

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134 / URL BLOWDOWN MODEL

CON	4	2	3	4	5	0	2	0	6	10	11	12	51	14	15	67	11	18	61	2.0	17	22	23	4.7	52	26	27	53	62	3.0
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COORDINATES DF X1-0R0 X1																														
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RST NODE THICK- X3-DRD NESS NUMBE	0*000 0*00	-*796 0.00	984 0.00	520 0.00	0.000 0.00	0.000 0.00	1.253 .0.00	1.838 0.00	2.373 0.00	3.020 0.00	3.667 0.00	4.130 0.00	4.130 0.00	4.130 0.00	4.130 0.00	4.130 0.00	4.130 0.00	4.130 0.00	4.130 0.00	.179 0.00	3.625 0.00	3.545 0.00	3.305 0.00	3.010 0.00	2.705 0.00	2.455 0.00	2.353 0.00	0.000 0.00	1.830 0.00	1 20.4 0.00
OF THE FIRST NODE THICK- X2-DRD X3-DRD NESS NUMBE	17.300 0.000 0.00	17.300796 0.00	17.300984 0.00	17.300520 0.00	17.300 0.000 0.00	*085 0.000 0.00	17.300 1.253 .0.00	17,300 1,838 0.00	17.300 2.373 0.00	17.300 3.020 0.00	17.300 3.667 0.00	16.765 4.130 0.00	15.834 4.130 0.00	14*903 4*130 0.00	13.971 4.130 0.00	13.040 4.130 0.00	12.550 4.130 0.00	12.209 4.130 0.00	12.109 4.130 0.00	0.000 .179 0.00	11.500 3.825 0.00	11.500 3.545 0.00	11.500 3.305 0.00	11.500 3.010 0.00	11.500 2.705 0.00	11.500 2.455 0.00	1:,500 2.353 0.00	034 0.000 0.00	11.500 1.830 0.00	11.600 1.208 0.00
COORDINATES OF THE FIRST NODE THICK- X1-ORD X2-ORD X3-ORD NESS NUMBE	0,000 17,300 0,000 0,00	-1.379 17.300796 0.00	-1.703 17.300984 0.00	-2.506 17.300520 0.00	-2.506 17.300 0.000 0.00	1.170 .085 0.000 0.00	-2.170 17.300 1.253 .0.00	-1.832 17.300 1.838 0.00	-1.832 17.300 2.373 0.00	-2.206 17.300 3.020 0.00	-2.579 17.300 3.667 0.00	-2.846 16.765 4.130 0.00	-2.845 15.334 4.130 0.00	-2,846 14,903 4,130 0,00	-2.846 13.971 4.130 0.00	-2.846 13.040 4.130 0.00	-2.846 12.550 4.130 0.00	-2.846 12.209 4.130 0.00	-2.546 12.109 4.130 0.00	431 0.000 .179 0.00	-3.374 11.500 3.825 0.00	-3.359 11.500 3.545 0.00	-4.275 11.500 3.305 0.00	-4.730 11.500 3.010 0.00	-5.314 11.500 2.705 0.00	-5.747 11.500 2.455 0.00	-5.920 1:.500 2.353 0.00	3.250034 0.000 0.00	-6.230 11.530 1.830 0.00	-6 287 11 500 1 203 0 00
/TYPE COURDINATES OF THE FIRST NODE THICK- /TYPE X1-ORD X2-ORD X3-ORD NESS NUMBE	0/ 0.000 17.300 0.000 0.00	0/ -1.379 17.300796 0.00	0/ -1.703 17.300984 0.00	0/ -2.506 17.300520 0.00	0/ -2.506 17.300 0.000 0.00	7/ 1.170 .085 0.000 0.00	0/ -2.170 17.300 1.253 .0.00	0/ -1.832 17.300 1.838 0.00	0/ =1.832 17.300 2.373 0.00	0/ -2.206 17.300 3.020 0.00	0/ -2.579 17.300 3.667 0.00	0/ -2.846 16.765 4.130 0.00	0/ -2.845 15.334 4.130 0.00	0/ -2.846 14.903 4.130 0.00	0/ -2.846 13.971 4.130 0.00	0/ -2.846 13.040 4.130 0.00	0/ -2.846 12.550 4.130 0.00	0/ -2.846 12.209 4.130 0.00	0/ -2.546 12.109 4.130 0.00	6/431 0.000 .179 0.00	0/ -3.374 11.500 3.825 0.00	0/ -3.359 11.500 3.545 0.00	0/ -4.275 11.500 3.305 0.00	0/ -4.790 11.500 3.010 0.00	0/ -5.314 11.500 2.705 0.00	0/ -5.747 11.500 2.455 0.00	0/ -5.920 1:.500 2.353 0.00	6/ 3.250034 0.000 0.00	0/ -6.230 11.500 1.830 0.00	A7

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0	-1.7030	17.3000	9840	0.00			
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0	-2.5060	17.3000	0.0000	0.00			
	-2.4210	17.3000	.6500	0.00			
2	-2.1700	17.3000	1.2530	0.00			
	-1.8320	17.3000	1.8380	0.00			
0	-1.8320	17.3000	2.3730	0.00			
22	-2.2060	17.3000	3.0230	0.00			
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9	-2.8460	14.9030	4.1300	0.00			
83	-2.8460	13.9710	4.1300	0.00			
0	-2.8460	13.0400	4.1300	0.00			
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2	-3.0007	11.6790	4.0405	0.00			
0	-3.3740	11.5000	3.8250	0.00			
0	-3.8590	11.5000	3.5450	0.00			
2	-4-2750	11.5000	3.3050	0.00			
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	~5.3140	11.5000	2.7050	0.00			
.0	-5.7470	11.5000	2.4550	0.00			
3	-5.9200	11.5000	2.3550	0.00			
7	-0.1465	11.5000	2.1272	0.00			
0	-6.2300	11.5000	1.8300	0.00			
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4ATERIAL PROPERTIES

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ELASTIC 430JLUS	*1987E*12 *1987E*12 *1987E*12 *1987E*12 *1987E*12 *1987E*12 *1987E*12
MEIGHT	<pre>. IO 375+06 . II 235+06 . I 2626+06 . 10916+06 . 21186+05 . 96106+05 . 11995+06</pre>
TEMPERATURE	
MATERIAL DESCAIPTION	CPIPE SPIPE SPIPE CPIPE VALVE SPIPE SPIPE
1ATERIAL NUMBER	

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# PIPE SECTION PROPERTIES

# HOR / UKL BLONDOWN MODEL

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	I 0 N A L 8 E N 0 X3-0x0 Y3-0x0 /3EN0 /FACT38 3A3115 (P,4)	0.,00 .53500000000101	-1.97 17.30	0.00 2.51	2.505¢¢¢¢¢¢¢¢[P]	*00 T(*30	+5*- 0°*0	-2.30 17.30			1.49 0.00								4.65	.61000000000010.	-3.37 12.11					2.7461	. b10sessesses. P1	-5.52 11.50		15.5	6.200******************	4.5061	.6100000000000000	-3.37 12.11		2.94	-5.62 LL**********
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	E N E R LEVEL/ ER INC	0 1	COORDINA 0 1	• •	PODDO THE	0 1	1 0	COORD INA	1 0	0	1	CODROIMA	0 1	0 1	1						CUUKU INA	1 .	4 -	1	1 0	0 1		CODRUINA	0	•	CODRD TNA	0 1		CODROINA		-	COURDINA
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### CONCENTRATED FORCES (LOAD CASE 1)

HDR / UKL BLOWDOWN MODEL

PAGE 10-24 MAR 81

### LOAD CASE 1 --

 NODE
 SPEC
 SKEW REST- CONS- REPEAT
 (X,R OR X*1 (Y,S OR Y*) (Z,T OR Z*)
 (X,R OR X*1 (Y,S OR Y*) (Z,T OR Z*)
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P16E 12 24 1.48 01 NODE SPEC SKEW REST- CONS- REPEAT (X,* OR X*) (Y,S OR Y*) (Z,T OR Z*) (X,R OR X*) (Y,S OR Y*) (Z,T OR Z*) *GENERATION* NUMBER SYST NODE RAINT TRAINT NODE FORCE FORCE FORCE FORCE ADRENT MOMENT MOMENT MOMENT NUMBER INC COUNT 21 FORCES (LOAD CASE BLOWDOWN MODEL CONCENTRATED HD& / UXL ; LOAD C156 2

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CUNCENTRATED FORCES (LOAD CASE 3)

MD4 / URL BLOWDOWN MODEL

PAGE 14 24 Max 81

LOAD CASE 3 --

NOUE SPEC SKEW REST- CONS- REPEAT (X+R GR X#) (Y+S DR Y#) (Z+T DR Z#) (X+R JR X#) (Y+S GR Y#) (Z+T DR Z#) #GENERATION* NUMBER SYST NODE RAINT TRAINT NODE FORCE FORCE FORCE FORCE MOMENT MOMENT NOMENT NUMBER INC COUNT -0 •0 •0 *0 .0 .10000E+01 0. -0 6.5

### CONCENTRATED FORCES (LOAD CASE 4)

HOR / URL BLOWDOWN MODEL

PAGE 15 24 MAR 81

LOAD CASE 4 --

NODE SPEC SKEW REST- CONS- REPEAT (X.R DR X*) (Y.S OR Y*) (Z.T DR Z*) (X.R DR X*) (Y.S OR Y*) (Z.T DR Z*) (X.R DR X*) (Y.S OR Y*) (Z.T DR Z*) *GENERATION* NODE RAINT TRAINT NODE FORCE FORCE FORCE NDMENT MOMENT MOMENT NUMBER INC COUNT 37 5 . 0. 0. 0. 0. 0. 0 1 1

CUNCENTRATED FURCES (LUAD CASE 5)

HDR / URL BLOWDOWN MODEL

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LOAD CASE

PAGE 13 24 MAK 31 SKEW REST- CONS- REPEAT (X,R OR X*) (Y,S OR Y*) (Z,T OR Z*) (X,R OR X*) (Y,S OR Y*) (Z,T OR Z*) *GENERATION* NODE RAINT TRAINT NODE FORCE FORCE FORCE FORCE MOMENT MOMENT MOMENT NUMBER INC COUNT SPEC NUDE

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10 FLUAD CASE FORCES 0 sta kon ** 1 1 3 3 N D 1

BLOWDOWN MODEL HOR / URL

P16E 24 MAX 61

ł LOAD CASE 6

*GENERATION* NUMBER INC COUNT NODE SPEC SKEW REST- CONS- REPEAT (X,R OR X4) (Y,S OR Y4) (Z,T O2, Z4) (X,R O2 X4) (Y,S U2 Y4) (Z,T O2 Z4) NUMBER SYST NODE RAINT TRAINT NODE FORCE FORCE FORCE FORCE FORCE PORCE MOMENT MOMENT MOMENT MOMENT đ 81

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HDR / URL BLOWDOWN MODEL

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PAGE 22 24 MAR 01

LOAD CASE 7

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----SKEW REST- CONS- REPEAT (X,R DR X*) (Y,S DR Y*) (Z,T DR Z*) (X,R DR X*) (Y,S DR Y*) (Z,T DR Z*) *GENERATION* NODE AAINT TRAINT NODE FORCE FORCE FORCE MOMENT MOMENT NUMBER INC COUNT -0 *0 •0 • 0 • .10000E+01 *0 SPEC 4 NUMBER 18

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CONCENTRATED FORCES (LOAD CASE 8)

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PAGE 24 24 MAR 61

LUAD CASE 8 ---

SKEW REST- CUNS- REPEAT (X,R UR X*) (Y,S UR Y*) (Z,T UR X*) (X,Z UR X*) (Y,S UR Y*) (Z,T UR 24) * GENERATION* NODE RAINT TRAINT NUDE FORCE FORCE FORCE FORCE MUMENT MUMENT NUMBER INC COUNT NOVE SPEC

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CUNCENTRATED FORCES (LOAD CASE 9)

HOR / URL BLONDOWN NODEL

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LOAD CASE 9 ---

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SKEW REST- COUSS- REPEAT (X+R OR X0) (Y+S DR Y0) (Z+T DR Z0) (X+R DR X01 (Y+S OR Y0) (Z+T DR Z0) 0GENERATION0 NODE RAINT TRAIN? NODE FORCE FORCE FORCE FORCE NUMENT MOMENT ADMENT NUMBER INC CUUNT NUMBER SYST

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#### CONCENTRATED FORCES (LOAD CASE 10)

HDR / URL BLOWDOWN MODEL

PAGE 23 24 MAR 81

LOAD CASE 10 --

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 SKEW REST- CONS- REPEAT (X.R OR X*) (Y.S OR Y*) (Z.T DR Z*) (X.R OR X*) (Y.S OR Y*) (Z.T DR Z*)
 (Y.S OR Y*) (Z.T DR Z*) (Z.T DR Z*)

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#### STNAMIC JOB

HOR / URL BLOWDOWN MODEL

PAGE 29 24 MAR 81

## TIME HISTORY ANALYSIS CONTROL VARIABLES

HOR / URL BLOWDOWN MODEL

NUMBER OF SOLUTION TIME STEPS - ( 5000) IDT (CC 15-20) SOLUTION TIME STEP .20000E-03) FDT (CC 21-30) TIME AT THE START OF THIS SOLUTION = ( -.96600E-011 FTIMED (CC 31-40) METHOD TO BE USED FOR DIRECT INTEGRATION . ( N) SN (55 41) EQ. N. NEHMARKAS METHOD EQ. A. WILSON THETA METHOD VALUE OF ALPHA PARAMETER (NEWMARK) - ( .50000E+00) FA (CC 42-50) VALUE OF BETA PARAMETER (NEWMARK) a 1 .250002+001 FB (00 51-00) VALUE OF THETA PARAMETER (WILSON) a ( 0. ) FT (CC 61-70) ANALYSIS TYPE (TRANSIENT OR STEADY-STATE) = ( T) ST (CC 71) PERIOD (STEADY-STATE ANALYSIS ONLY) = ( 0. ) FTL 106-57 221

*** E R R O R ** PRECEDING LINE. SECTION (HISTORY), CARD(, 1), COLUMNS (31-40). ZERO (OR NEGATIVE) CONTROL PARAMETER.

PROCESSING TERMINATED FOR THIS SECTION.

PEGE 30 24 MAR 81

#### JAMPING RATIOS

HDR / URL BLONDOWN MODEL

PAGE 31 24 MAR 81

				~ ~
MASS PROPORTIONAL DAMPING CONSTANT = ( (DIRECT INTEGRATION OPTION ONLY)	.12000∃+01}	FALPHA	(00	31-40)

STIFFNESS PROPORTIONAL DAMPING CONSTANT = ( .63500E-04) FBETA (CC 41-50) (DIRECT INTEGRATION OPTION ONLY) APPENDIX B

PRESSURE WAVE FLUID FORCES ON URL PIPE

#### APPENDIX B

## PRESSURE WAVE FLUID FORCES ON URL PIPE

The approach used to determine the fluid forces on the URL pipe was to assume the value of the momentum terms in the linear momentum equation, from fluid mechanics (control volume formulation), to be small compared to the surface force terms. For this reason, the momentum terms, together with the body force term, were neglected for this analysis. The resulting equation to be solved for the fluid force on the pipe  $\overline{F}_R$  is  $\overline{F}_s = 0$ . The solution is obtained simply as follows (see Figure B.1):

$$\overline{F}_{s} = -\overline{F}_{R} + p_{1}A_{1}\hat{i}_{1} - p_{2}A_{2}\hat{i}_{2} = 0$$
 (sum of the surface forces on the C.S.)

$$\overline{F}_{R} = p_{1}A_{1}\hat{i}_{1} - p_{2}A_{2}\hat{i}_{2}, \qquad (B-1)$$

where  $\overline{F}_{R}$  is the resultant force of a control volume on its corresponding section of pipe. This equation is applied, as follows, to the control volumes defined for this solution approach.



FIGURE B.1: Pipe Force on Control Volume.

SHEET ____ OF 4 DESCRIPTION GSP4a ANCO Engineers, Incorporated ANCO 1701 Colorado Avenue, Santa Monica, CA 90404 (213) 829-9721, 829-2624 2/21/81 CALCULATIONS FOR 1182-8 (NRC) MADE BY _ WBW DATE _ 9/4/4/ CHECKED BY ____ GEH DATE

Force of Fluid ( Control Volumes) On Pipe Apply equation B-1 as follows (use Figures B.2-B.7): F = P, A, Sin 29° (- i) + P, A, cos 29° j 17 : + BAZi = (BA2 - P, A, sin 29) i + P, A, cos 290 f

 $\overline{F_2} = -P_2 A_2 \hat{i} + P_3 A_3 \hat{i}$ LP2 :  $= (P_3 A_3 - P_2 A_2)\hat{i}$ 

 $LP_{3}: \bar{F}_{3} = P_{4}A_{4}\hat{j} + P_{5}A_{5}\hat{i}$  $= P_{5}A_{5}\hat{i} + P_{4}A_{4}\hat{j}$ 

 $LP_{4}: \bar{F}_{4} = P_{c}A_{6}\hat{k} + P_{A}A_{7}\cos 60^{\circ}\hat{i} + P_{A}A_{7}\sin 60^{\circ}\hat{j} + P_{A}A_{7}\sin 60^{\circ}\hat{j} + P_{A}A_{7}\sin 60^{\circ}\hat{j} + P_{A}A_{6}\hat{k}$   $= P_{A}A_{7}\cos 60^{\circ}\hat{i} + P_{A}A_{7}\sin 60^{\circ}\hat{j} + P_{6}A_{6}\hat{k}$ 

SHEET & OF 4 ANCO Engineers, Incorporated DESCRIPTION GSP44 ANED 1701 Colorado Avenue, Santa Monica, CA 90404 (213) 829-9721, 829-2624 DATE 7/21/81 CALCULATIONS FOR 1182-8 (NRC) WADE BY WBW DATE 9/11/81 CHECKED BY

F5 = - P, A, Los 15° i - P, A, sin 15° j LP=:

These equations were implemented in solving GSP4a. They were coded (programed) in the compation code PNFORC. A listing of this code is firm at the such of this appendix.

SHEET _3 or 4 DESCRIPTION GSP40 ANCO Engineers, Incorporated ANED 1701 Colorado Aver. vo, Santa Monica, CA 90404 (213) 829-9721, 829-2624 CALCULATIONS FOR 1182-B (NRC) WBW DATE _7/21/81 MADE BY CHECKED BY DATE



* Note: See Figure B. 7 concerning the definition of the unit vectors i, j, k. B-6

+8,700

BHEET _ 4 00 4 ANCO Engineers, Incorporated GJP4a ANCO DESCRIPTION 1701 Colorado Avenue, Santa Monica, CA 90404 (213) 829-9721, 829-2624 ____ DATE 7/21/81 MADE BY __WBW CALCULATIONS FOR _1182-8 (AURC) CHECKED .... DATE 756 Figure 8.4 K LB -11,820 2 1080 60 - 10,420 0 718 Fig 4121967 8.5 Figure B.6 967 3500 LP4 14.38 535 LPA 35 PGAG 3811

PGA6 S

Figure B.7

LOCAL COORDINATE DIRECTIONS FOR APPLIED FLUID FORCES



LP, = load point j (corresponds to control volume C.V.j) [] = bracket pair designates a control volume

Commentation of the pressure with the pressure with the pressure of the pressure with the pressure of the pressure with the pressure of the pr PROGRAM PWFORCELNPUT, OUTPUT, TAPES+INPUT, TAPE6+OUTPUT, TAPE7, TAPE8; CALCULATE FORCES OF FLUID ON UALIBDI PIPE(PRESSURE WAVE OWLYP FORCE INFORMATION *CALL SUBROUTINE DOSYSR TO READ GERMAN DATA IN DOSYS FORMAT CALL DOSYSGTIME, P.50000, 7.1 WHERE, NTPTS, 81 DIMENSION PI7,50001.4171.FI10.50001.TITLE1401 0146NSTON TIME(50001.14HERE:101.0T150001 DIMENSION P0(71 ** CALC. AVERAGE TIME INCREMENTIAJTI** LP CODRD. DIRECTION DIF = ABS(0T(1) - ADT) IF(DIF.GT.DIFMAX) DIFMAX = DIF ** COMPAIR DIGII WITH ADT ** 01111 - 11-11 - 11WELL + 11 - 11WELL (L)04 - (TJ.L)4 - (TJ.L)4 F08CE AT LP F11.11 F12.11 * READ INPUT DATA **
 * READ DATA FROM CARDS READ(5,10) TITLE
 READ(5,11) NIPTS BIAS THE PRESSURES
 BIAS THE PRESSURES
 PO(J) PP(J) P(J)
 PO(J) P(J)
 PO(J) P(J)
 PO(J) P(J) READ(5,412) A READ(5,400) 14HERE FORMAT(1015) ADT - 401 - 0111 I - STOIN - IMM 3 CON ANCO *** 1PM.1=1 06+ 00 1MN+1+1 01+ 00 00 +20 I-1,NML DIFMAX -1.0 POLNT 11.91 RE41ND 7 RE41ND 8 ADT - 0.0 LOAD CONTINUE 10.00 +00 666 0001 01+ 420 430 44 000 U. 00 0000000000 000

B-9

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F14,11
          - 14
                37
                                 14
                37
          3
                     F15,11
                                 Y
                     F16+11
                18
                     F17.11
                18
                18
                     F(8,1)
                    F(9,1)
                11
                11 F(10,1)
     . FORCES F(1,1) THRU F(10,1) ARE COMPONENTS IN LOCAL COORD. SY5.5
    · PRESSURE INFORMATION
      ANCO
       NODE
             PRESSURE AREA(M##2)
              P(1,1) A(1) = 0.1083
       51
       46
              P12.11
                      A(2) = 0.1083
       43
              P13.11
                       A(3) = 0.2083
                       A141 = 0.1083
       40
              P14.11
                       A(5) = 0.1018
       36
              P15,11
       22
              P16+11
                      A(6) = 0.1018
              P17.1) A171 = 0.1018
       14
      (NOTE-- IN THE ABOVE - I - IS INTEGER TIME.)
      (NOTE--ANCO NODES(NO.S) ARE THOSE FROM THE EASE2 MODEL FEBBI)
     * THE AREA A IN THE EQ.S FOR F IS ACTUALLY A* = 10**5*A(#**2)
      AND HAS UNITS N/BAR.
     . THE UNITS OF P ARE BAR.
     . THE UNITS OF F ARE N.
     . INITIALIZE ARRAY F
    NTPTSS = (INT((NTPYS-0.5)/8)+1)*8
    00 300 K=1,10
    DO 300 1=1,NTPT55
300 F(K,1) = 1.0
     . CALC. ARRAY F
    00 100 IT-1.NTPIS
    FI1,IT) = PI2,IT)+A(2) - 0.485+PI1,IT)+A(1)
    F12.1T1 = 0.875*P(1.1T)*A(1)
    F13,11) # P13,113#A13) - P12,113#A12)
    F14.111 + P15.111*A(5)
    F15.111 = P(4.11)+A(4)
    F16.11) = 0.500*P17.11; *A(7)
    F(7.11) = 0.866*P(7.11)*A(7)
     F(8.11) = P(8.11)*A(6)
    F(9,11) = -0.966*P(7,11)*A(7)
     F(10,1T) = -0.259*P(7,1T)*A(7)
100 CONTINUE
     ** END CALCULATING FORCES **
     .. WRITE OUT DATA ..
     . WRITE THE FORCES OUT TO TAPETIEASEZ FORMATI
     LTAPE . 7
     00 150 K=1,10
     ARITEILTAPE,2071 (FIK,1),1+1,NTPTS5)
     ENDEILE LTAPE
150 CONTINUE
     . WRITE OUT TO OUTPUT
     4RITE16,2001
     WRITEI6,2011 TITLE
     #RITE(6.2021 NTPTS
     WRITE(5,440) TIME(1),ADT.DIFMAX
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10 F08matt40a2)
11 F03matt4015)
12 F08matt40105)
200 F08matt401475
200 F08matt414747960cam PwF08C -- Calculates The PressJae wave .
201 F08matt41 + 27HE00 wrl Pite F1.5) = ,15,//)
202 F03matt41 + 27HE00 AREAS(A1A2....A7) = ,75L2.3,//)
203 F03matt41 + 27HE00 AREAS(A1A2....A7) = ,75L2.3,//)
203 F03matt41 + 27HE00 AREAS(A1A2....A7) = ,75L2.3,//)
204 F03matt41 + 24HE5500 E TIME H1ST0AY F(.11,14),//)
205 F03matt41 + 24HE5500 E TIME H1ST0AY F(.12,14))
205 F03matt41 + 24HE50 F1ME H1ST0AY F(.12,14))
205 F03matt41 + 24HE708 = 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 100 + 
440 FORMATI 1X.10HADDITIONAL IMFG..//.1X.

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8 BHITEG.203 A

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01YENSION TIME MAXPT) PRESSRINPSR.MAXPT1, IMMERE(MP31, DIWPUT DIMENSION TIME MAXPT) PRESSRINPSR.MAXPT1, IMMERE(MP31, DIWPUT BEAD 305Y5 HEADER INFORMATION AND DUTPUT TO PRINTER(CHAN 6) READICHAN.9031) USERID READICHAN.9031) USERID READICHAN.9021) HOR READICHAN.9021) HOR READICHAN.90210 HOR READICHAN.90210 HOR READICHAN.90210 HOR READICHAN.90210 DESCRP DO 100 1-7,10 READICHAN.90270 CONTINUE READICHAN.90271 NTRAN READICHAN.90271 NTRAN READICHAN.90271 NTRAN READICHAN.90271 NTRAN READICHAN.90217 NTRAN				
DITENSION THEIMAXPTL,PRESSMMPSR,MAXPTL,MHERE(MPSR,DINPUT DIMENSION TSTRUM(2),DESCRP(3),JL4EEL(2),ICOORD(3),ISTAT(3) READ 305Y5 HEADER INFORMATION AND DUTPUT TO PRINTER(CHAN 6) READ(ICHAN,9001) USERID WRITE(6 *9101) USERID WRITE(6 *9101) USERID WRITE(6 *9102) HOR WRITE(6 *9103) ISTRUM WRITE(6 *9103) ISTRUM WRITE(6 *9103) ISTRUM WRITE(6 *9103) ISTRUM WRITE(6 *9103) ISTRUM WRITE(6 *9101) IDATE READ(ICHAN,9003) DATE READ(ICHAN,9003) DATE READ(ICHAN,9003) DATE READ(ICHAN,9003) DATE READ(ICHAN,9003) DATE READ(ICHAN,9003) DESCRP READ(ICHAN,9003) DESCRP READ(I		********	***********************************	11500********************
READ JOSYS HEADER INFORMATION AND DUTPUT TO PRINTERICHAN 6) READICHAN.9011 USERID WRITEL6 .91011 USERID READICHAN.90011 USERID READICHAN.90021 HOR READICHAN.90021 HOR READICHAN.90031 TSTRUM RETEL6 .91031 TSTRUM READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90041 TSTRVP.ICDDE.MODEL.ITYPE.IDAMP READICHAN.90011 TDATE READICHAN.90051 DESCRP REITEL6 .91011 IDATE READICHAN.90071 CONTINUE READICHAN.90071 CONTINUE READICHAN.90171 NTRAM READICHAN.90171 NTRAM RETELCA .91071 NTRAM READICHAN.90171 NTRAM READICHAN.90171 NTRAM READICHAN.90171 NTRAM		12N3K10	STON TIME (MAXPT) . PRESSALNPSR . MAXPT) . IMES	ELNPSAI, DINPUT (2561035YL)
READ JOSYS HEADER INFORMATION AND DUTPUT TO PRINTERICHAN 51 READICHAN-9011 USERID WRITEL6 -91011 USERID READICHAN-90021 HOR WRITEL6 -91021 HOR READICHAN-90021 HOR READICHAN-90021 HOR READICHAN-90021 HOR READICHAN-90031 TSTRUM READICHAN-90031 TSTRUM READICHAN-90031 TSTRUM READICHAN-90031 DATE READICHAN-90031 DATE READICHAN-90031 DATE READICHAN-90031 DESCRP READICHAN-90031 DESCRP READICHAN-90031 DESCRP READICHAN-90031 DESCRP READICHAN-90031 DESCRP READICHAN-90031 CONTINUE READICHAN-90171 NTRAM READICHAN-90171 NTRAM READICHAN		1000000	STOR TATEORIELANCONCLASSICASELICIAN	11000 101101010101010101010101010101010
READIICHAN.9001) USERID WRITE(69101) USERID KEADIICHAN.90021 HOR KEADIICHAN.90021 HOR KEADIICHAN.90021 TSTRUM KEADIICHAN.90031 TSTRUM KRITE(691031 TSTRUM KRITE(69101) IDATE KEADIICHAN.90041 TSTRPF.ICDDE.MDDEL.11TYPE.IDAMP KEADIICHAN.90051 DESCRP KEADIICHAN.90051 DESCRP READIICHAN.90051 DESCRP READIICHAN.90071 CONTINUE READIICHAN.90071 CONTINUE READIICHAN.90171 NTRAM KEADIICHAN.90171 NTRAM KRITE(691071 NTRAM KRITE(6 .		READ DO	JOSYS HEADER INFORMATION AND DUTPUT TO PI	INTERICHAN 51 JOSY13
READITCHAN, YOOT) USEKID READITCHAN, YOOT) USEKID READITCHAN, 9002) HOR READITCHAN, 9002) 157KUM WRITE(6, 9102) 157KUM READITCHAN, 9003) 157KUM READITCHAN, 9004) 157TYP, ICODE, MODEL, ITYPE, IDAMP READITCHAN, 9004) 157TYP, ICODE, MODEL, ITYPE, IDAMP READITCHAN, 9004) 104TE READITCHAN, 9005) 055CRP WRITE(6, 9101) 104TE READITCHAN, 9007) 055CRP WRITE(6, 9101) 104TE READITCHAN, 9007) CONTINUE READITCHAN, 9007) READITCHAN, 9007) CONTINUE READITCHAN, 9017) NTRAM READITCHAN, 9017) NTRAM				CLASEC
READITCHAN.9027 HDR WRITE(6 .9102) HDR READITCHAN.90021 157RUN READITCHAN.90021 157RUN READITCHAN.90041 157TYP.ICD0E.MDDEL.ITYPE.IDAMP READITCHAN.90041 10ATE READITCHAN.90041 10ATE READITCHAN.90051 DESCRP READITCHAN.90051 DESCRP READITCHAN.90051 DESCRP WRITE(6 .9102) 10ATE READITCHAN.9007) CONTINUE READITCHAN.9007) CONTINUE READITCHAN.9007) CONTINUE READITCHAN.90171 NTRAN READITCHAN.90171 NTRAN RETECLO .91071 NTRAN READITCHAN.90171 NTRAN		WEAUTIC IS	ICMAN, YUUIJ USEKID	CTA 500
WRITE(6 *9102) HDR KEADI(CHAN,9003) ISTRUM KEADI(CHAN,9003) ISTRUM KEADI(CHAN,9003) ISTRUM KEADI(CHAN,9003) ISTRUM KEADI(CHAN,9001) IDATE KEADI(CHAN,9001) IDATE KEADI(CHAN,9005) DESCRP WRITE(4. 9105) DESCRP WRITE(4. 9105) DESCRP READI(CHAN,9007) CONTINUE READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,9007) READICHAN,900707) READICHAN,9007) READICHAN,9007) READICHAN		READITC	CHAN. 90321 HOR	
READIICHAN.9003) ISTRUM WRITE(69103) ISTRUM READIICHAN.9003) ISTITP.ICD0E.MDDEL.IITYPE.IDAMP ARITE(59103) ISTITP.ICD0E.MDDEL.IITYPE.IDAMP ARITE(69103) IDATE WRITE(69103) DESCRP WRITE(49103) DESCRP WRITE(49103) DESCRP DO 100 1=7.16 DO 100 1=7.16 CONTINUE CONTINUE READIICHAN.9007) CONTINUE READIICHAN.9007) CONTINUE READIICHAN.9007) NARAN READIICHAN.9017) NTRAN READIICHAN.9017) NTRAN READIICHAN.90170 NTRAN READIICHAN READII READIICHAN.90170 NTRAN READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READII READI		WRITEIS	16 .91021 HDR	*1AS00
<pre>KETTEL6 .91031 ISTRUN READICHAN,90041 ISTTYP.ICD0E.MD0EL.ITYPE.IDAMP ARTEL6 .91041 ISTTYP.ICD0E.MD0EL.ITYPE.IDAMP KEADICHAN,9001) IDATE WRITEL6 .9101) IDATE READICHAN,90050 DESCRP WRITEL4 .91050 DESCRP WRITEL4 .91050 DESCRP DO 100 I=7.16 READICHAN,90070 Continue READICHAN,90070 READICHAN,90070 READICHAN,90070 READICHAN,90170 NEAN READICHAN,90170 NEAN READICHA</pre>		READIIC	ICHAN. 90031 TSTRUM	114500
KENTICHAN, 900-1 TSTTTP.ICODE.MDDEL.ITTTPE.IDAMP KENTIELS 910-1 TSTTTP.ICODE.MDDEL.ITTTPE.IDAMP READILCHAN, 9001 JOATE WRITELS 9101 JOATE READILCHAN, 9005 DESCRP WRITELA. 9105 DESCRP MRITELA. 9105 DESCRP DO 100 I=7,16 READICHAN, 9007 DESCRP OD 100 I=7,16 READICHAN, 9007 DESCRP READICHAN, 9007 DESCRP READICHAN, 9007 DESCRP READICHAN, 9007 DESCRP READICHAN, 9017 NTRAN READICHAN, 9017 NTRAN READIC		WRITE16	16 .91031 TSTRUN	1ASCC
READICHAN, 9001) TOTT FROM CONTRACTION FROM FROM FROM FROM FROM FROM FROM FROM		ALADI IC	ICMAN, 40041 ISITYP, ICODE, MODEL, ITYPE, IDA.	LASEC d
WRITEIG		8FADITCIO	ICHAN. GODIN TOTICYLGUDENDUEL MUTERIUM	
READTICHAN.9006) DESCRP           WRITE(6		WRITELS	16 .91010 IDATE	TASED
WRITE(4. +9106) DESCRP DD 100 1=7,16 DE 2001[CHAN+9007] C CONTINUE READICHAN-9007] READ TRANSDUCER IDENTIFICATION HEADER AND DUTPUT TO PRINTER READICHAN-90171 NTRAN ARITE(5 +91071 NTRAN ARITE(6 +91071 NTRAN READICHAN-90171 NSAMP WRITE(6 +91071 NSAMP REISATH-GT-MARPT) GOTO 1000 D0 200 1~1.WTRAN		READEIC	ICHAN. 90061 DESCRP	1300
DO 100 1-7.16 READICHAN-9007) C CONTINUE READ TRANSDUCER IDENTIFICATION HEADER AND DUTPUT TO PRINTER READICHAM-9017) NTRAN ARITEIS - 99107) NTRAN ARITEIS - 99107) NTRAN READICHAN-9017) NSAMP MRITEIS - 99107) NSAMP REISAR-GT-MARPT) GOTO 1000 DO 200 1~1.478AN		WRITEIG	14. +91061 DESCRP	14500
READIICHAN, 907) CONTINUE READ TRANSDUCER IDENTIFICATION MEADER AND DUTPUT TO PRINTER READIICHAN, 90171 NTRAN ARITEL6 .91071 NTRAN READIICHAN, 90171 NSAMP WRITEL6 .91071 NSAMP WRITEL6 .91071 NSAMP READIICHAN, 90171 NSAMP READIICHAN, 90171 NSAMP READIICHAN, 90171 NSAMP DO 200 1~1, MTRAN		00 100	1 1=7+16	14566
CUNIINUE READ TRANSDUCER IDENTIFICATION MEADER AND DUTPUT TO PRINTER READLICHARD,90171 NTRAN ARITEIS -91071 NTRAN READLICHAN,90171 NSAMP READLICHAN,90171 NSAMP REIECS -91071 NSAMP IFINSARY-GT-MARPT JGGTO 1000 DO 200 1×1+NTRAN		READI	01 ICHAN, 90071	TASEC
READ TRANSDUCER IDENTIFICATION HEADER AND DUTPUT TO PRINTER READITCHAM-90171 NTRAN ARTELS -91071 NTRAN READITCHAN-90171 NSAMP READITCHAN-90171 NSAMP IF(NSAMP-GT-MARP13 GOTO 1000 DO 200 I=1.NTRAN	3	CONTINU	406	LASED
READIICHAM-90171 NTRAN ARITELS - 91071 NTRAN READIICHAN-90171 NSAMP MRITELS - 91077 NSAMP IFINSAMP-GT-MAXPT1 GOTO 1000 DO 200 1~1.MTRAN		READ TR	FRANSDUCER IDENTIFICATION HEADER AND DUTI	TASEC ABINING OT TU
READIICHAM-99171 NTRAN ARITE16 - 91071 NTRAN READIICHAN-90171 NSAMP MRITE16 - 91071 NSAMP IF(NSAMP.GT-91071) GOTO 1000 DO 200 1~1.WTRAN				514560
#KITE(b. +910/1 MIKAN READITCHAN,9017) NSAMP WRITE(b. +9107) NSAMP IF(NSAMP.GT.MAXPT) GUTO 1000 DO 200 1ml.WTRAN		READLIC	ICHAR, 901 71 NTRAN	517SCC
WRITEI6 . 91077 NSAMP IFINSAMP.GT.MAXPT1 GOTO 1000 DO 200 1×1.WTRAN		READLIC	ICHAN. ODI 71 NCAMP	1A3LC
IFINSAMP.GT.MAXPT1 GUTO 1000 DO 200 1.41.MTRAN		WRITEIS	(6 ,9107) NSAMP	
DO 200 1-1.WIMAN		IF (NSAM	AMP. 67. MAXPTI 6010 1000	CTASCE
		00 200	0 I "I "NTRAN	91,4500
READIICHAN, 90191 INAME, TCODE, TPUC, TOIR, TLABEL, TSYS, ICOORD,		READI	DIICHAN, 90191 INAME, TCODE, TPUC, TOIR, TLAB	L.TSYS.ICOORD.JCOJEJJSY16
		ST E MR	ITED STATES IMMORFLANCESITUUSIUSIUSI	TALASSAN MALANANA SAELSINA

04914500 04914500 04914500	07214500 001214500 000214500 00914500 00914500	0641450C 0921450C 0921450C 0921450C 0921450C 0941450C	02874500 00874500 00874500	0/614500 02/814500 05/814500 05/814500 05/814500	01074500 00074500 05614500 05614500 05614500 05614500 05614500 05614500
RFADIICHAN,90201 TSTAT WRITTIG +91201 TSTAT CONTINUE	READ INPUT DATA AND SAVE REQUESTED PRESSURES DON'T READ PRESSURES IF IMMERELII=0. If LIMMERELII.E0.01 GOTO 500 DO 400 I=L.NSAMP READTICHAN.90211 TIMELLI.fDINPUTLJI.J=L.NTKANI	D0 300 J-1,MFSR PRESSR1J.[] = DINPUT(IMMERE(J)) CONTINUE CONTINUE NPTS = NSAMP RETURN	CULTURE CASOON STOP FORMATS	FORMATLIX*,31M UMBLE TO MANOLE NO. OF POINTS! FORMATLIA,74X) FORMATLAD,74X) FORMATLAD,42.68X! FORMATLAID,43.56X! FORMATLAID,44.56X! FORMATLAID,44.56X! FORMATLAID,44.56X!	FGRMAT(A6.42.46.41.410.44.43.315.41.324) FGRMAT(212.5.88) FGRMAT(212.5.88) FGRMAT(12.16) FGRMAT(12.46) FGRMAT(12.46) FGRMAT(12.41.43.21].12.16) FGRMAT(12.41.43.21].12.16) FGRMAT(12.13.43.21].12.10.44.33.315.41) FGRMAT(12.210.46) FGRMAT(12.210.46) FGRMAT(12.210.46) FGRMAT(12.210.46) FGRMAT(12.210.46)
200		300 500	000	9004 4004 4006 4006 4006 4006	9019 9020 90219 9103 9103 9107 9107 9107 9107 9107 9107 9107 9107

# APPENDIX C

PROCESSING OF OUTPUT FROM EASE2 AND COMPUTATION OF ADDITIONAL STRESSES

## APPENDIX C

# PROCESSING OF OUTPUT FROM EASE2 AND COMPUTATION OF ADDITIONAL STRESSES

The computer code PWSTRS was written to apply coordinate transformations to the displacement and acceleration output from EASE2, compute additional stress components, change to desired units (i.e., displacements were to have units of mm), and format solution results in DOSYS format. A listing of this code is given in this appendix.

11-0x11 11-0x111 11-0x1111 11-0x111 11-0x111 11-0x1111 11-0x111 11-0x111 Certes Process Provide DATA LTAPES.LTAPES/8.97 DATA TTAPE.TAPOUT/10.97 DATA TTAPE.TAPOUT/10.97 DATA ZAN/0.0.0.01974 .0.0.0.01974 .0.2259 / DATA ZAN/0.0.0.01974 .0.0.0.01974 .0.2259 / ADD FA.AND TF ARE THE COORD. TAANS. MATRICIES FOR THE DISPL.. ACCEL.. AND FORCE . RESPECTIVELY. DATA TD/0.510.66610.0.0000.0.100.0.100.0.8664.0.5.0.0.0. \$ E + C A ] } 12 * ON 1 1 FAPE10, FAPE11, FAPE12, FAPE13, FAPE14, FAPE15, FAPE15, FAPE18,
 FAPE19, FAPE20, FAPE21, FAPE22, FAPE23, FAPE24, FAPE25, FAPE25, FAPE27,
 FAPE26, FAPE29, FAPE30, FAPE311 11 PROGRAM PMSTRSILNPUT, OUTPUT, TAPES-ENPUT, TAPE6-OUTPUT, TAPEL. ** READ PRESSURE TIME HISTORIES * CALL SUBROUTINE DOSYSR TO READ GERMAN DATA IN DOSYS FORMAT CALL DOSYSRIPHTIME,P,5000,7,1MHERE,NTPTS,93 11.513815. 11 DITENSION NORS(7).SI(2.0).S2(2.0).S3(0).L3RN(6).X1(5) DIMENSION TTIME( 1).0X(3, 1).0Y(3, 1).0Z(3, 1) DITENSION XX(1, 1).AY(1, 1).AZ(1, 1) 11.54(5,5000) DITENSION AXII. 11.AYII. 11.AXII. 11.AXII. 11 DITENSION AXII. 11.AYII. 11.AXII. 11. DITENSION AYIS. 11.5FEN(S. 11.58EN(S. 1 DITENSION FXIS. 11.FYIS. 11.ALPA445. 11 DITENSION FXIS. 11.FYIS. 11.54(S. 11.977 DITENSION SPA15. 11.5FIG. 11.54(S. 11.577 DITENSION PAIRE(SOU) 11 INTEGER EZMIST INTEGER TIAPE, DTAPE, ATAPE, STAPE, ALTAPE, FTAPE 14PE8.14PE9. ó × DITENSION MEDI(8), HEDZ(8) 0 ú a × E2HIST/1/ REWIND E2HIST REWIND LTAPE8 REWIND LTAPE9 00 666 1+10+29 REWIND I 10 REAL MY . ML œ 1 2 0= 41 251 DATA 0 0 . . 660 000 00

0

23

0

4

0

31 Ji--9 CHARATER 1A91. LINE 3 OUTPUT PREFIX FOR ALL PMAXCOL*REQJEST QUANTITIES. NJTE THAT FOR THE CASE 1531JI.E0.9M TH-J-TH REQUEST QUANTITY IS AN ARTHMETIC COMPUTATION SEQUENCE INUMBERL3RN(JJ)*0.4ND SIL1.JI.SIL2.JI AND SZ[1.JJI.SZ[2.J] ARE THE ASS LINE L AND LINE 2 HEADINGS. IF THE ASS LINE L AND LINE 2 HEADINGS IF THE J-TH REQUEST IS NOT AN ACS QUANTITY. 531JI DEPENDS ON THE CATEGORY NUMBER (AET) AS GIVEN IN THE FOLLOWING TABLE HED1, HED2 * EASE2 TITLE CARDSFEACH BALD)
101 * NUMBER OF SOLUTION STEPS
FD7 * TIME STEP LINCAEMENT)
FD7 * TIME STEP LINCAEMENT)
NUMBS * NUMBER OF OUTPUT REQUEST SETS IN EACH JF THE SEVEN CATAGORIES (HODE FORC, BEAM, MEMB, STEL, SOLI, PIPEL AS DETENTINED BY THE DATA ENTRIES
READIEZMIST) HEDIMG AND PERTIMENT INFO. CATEGORY (1-NODE.2-FORC.3-BEAM....,7-PIPE) AN OUTPUT SET NUMBER IN THE KEY-TH CATEGORY (1.LE.MOS.LE.NORSKEY) SOLUTION TIME STEP INTERVAL AT WHICH JJ'°.T IS AWRITEN TO THE E2MIST FILE MAXIMUM NUMBER DF REQUEST GJANTITIES IN THE MAXIMUM NUMBER DF REQUEST GJANTITIES IN THE NGS-TH OUTPUT SET.KEY-TH CATEGORY PREFIX, 53(J) NODE . BEAM MEM3RANE 0017901 94 94 94 94 94 94 94 94 94 94 I.LE. MAXCOL.LE.81 CATEGORY NUMBER 1.2 EADER RECORD 1 FIEDFIE 2HISTI 1000.5 . . . 165 VARIABLES **WAXCOL** ITPH. KEY SON x 00000000000 13 0000

· LINE 3 REFERENCE NUMBER FOR THE J-TH REQUEST

LIRNILL

VARIABLES

```
QUANTITY
C
                     KEY .EQ.1.2
                                        . NODE NUMBER
C
                     KEY .EQ. 3, 4, 5, 6, 7 . ELEMENT NUMBER
C
                     531J1.EQ.9H
                                     ACS. ACS NUMBER
C
C.
    5 READIEZHISTIKEY, NOS. ITPH, MAXCOL,
                   (51(1, J), 51(2, J), J=1, MAXCOL),
                   (SZ(1, J), SZ(2, J), J=1, MAXCOL),
                   (S31 J).L3RN(J).J=1.MAXCOL)
       IFIEDFIEZHIST) 11060.80
   BO CONTINUE
       WRITE(6, 302) KEY, NOS . ITPH
  302 FORMATIIX, 24HINFO. FROM HEADER RECORD, //.
              9X, 20HCATEGORY NO. IKEYI = .15./ .
              9X, 32HOUTPUT SET NO. IN KEY-TH CAT. = ,15,/,
              9x.18HOUTPUT INTERVAL # +15./)
       #RITE16,3031 53(1).L3RN(1)
  303 FORMAT(9X.49.15.//)
       L . LIRNILI
       .. BRANCHING, DEPENDENT UPON TYPE OF DATA
C
       IFIKEY.EQ.11 GOTO 51
       IFIXEY.EQ.21 GOTU 53
       IFIKEY.EG. 71 GOTO 52
       ** READ AND SET UP NODE DATA FOR OUTPUTIDISPL.S AND ACCEL.SI
C
   51 CONTINUE
       IF(L.EQ. *** IND=1
       IFIL. 69. 2 1N0=2
       IFIL. 62.4211ND=1
       [F(L.E0.43)]ND=3
       LIND REFERRS TO THE INTERNAL NODE NUMBERS FOR OUTPUT OF NODE
C
       DATAIKEY=11.1
C
       #RITE(6,304) L.IND
  304 FORMATELX, 32HINFL. FROM COMP. SEQ. (NODE DATA) .//.
              PX. 20HEXTERNAL NODE NO. . . 15./.
     .
              9X.20HIN ERNAL NODE NO. # .15.//1
     .
       GOTO 10
       C
   52 CONTINUE
       IFIL.EQ.ININD=1
       IFIL.E0.711ND=2
       1F(L.E0.1511ND=3
       IFIL.E0.1711ND=4
       IFIL.60.2111N0=5
       LIND REFERRS TO THE INTERNAL NODE NUMBERS FOR DUTPUT OF PIPE
C
C
       JATALKEY=71.)
       WRITELS, 3051 L, IND
  305 FORMATILX, 32HINFO. FROM COMP. SEQ. (PIPE DATA).//.
              9X.ZOHEXTERNAL PIPE NO. . . 15./.
      ٠
              9X. ZOHINTERNAL NODE NO. # 15.//1
      .
       GOTO 10
       ... READ AND SET UP FORCE DATA FOR OUTPUT (FORCES)
C
   51 CONTINUE
        IFIL.EQ.11111N0-1
        1F(L.E2.18) IND=2
        1F(L.EQ. 37) IND =3
        IF(L.EQ.43) IND+4
        IFIL.EQ. 4911ND=5
        LIND REFERRS TO THE INTERNAL NGOE NUMBERS FOR OUTPUT OF FORCE
 C
        DATA(KEY=2).)
        #RITE16.305) L.IND
   306 FORMATELX. 33HINFO. FROM COMP. SEQ. (FORCE DATA).//
              9x. 20HEXTERNAL NODE NO. . . 15./.
      .
      .
               9x.20HINTERNAL NODE NO. = .15.//1
```

C-5

#### DEFINITION OF INTERNAL NODE NUMBERS(IND) (2/13/51)

IND REFERRS TO A NUMBERING SYSTEM WITHIN THIS CODE THAT CORRESPONDS TO AN EXTERNAL (PHYSICAL) NUMBERING SYSTEM OF NODE AND ELEMENT NUMBERS.

EXTERNAL		NODE DATA
NODE		CORRESPONDING
NUMBER	IND	RESPONSE DUANT
14	1	ALSPUISE WORKIS
12		UTSPETS-AN ANN KON-SIDITENI
22	2	DISPLIT. TH VON RDB-STUTZENI
42	1	ACCELIAM VENTIL)
43	3	DISPLIAN VENTILI
EXTERNAL*		ELEMENT DATA
ELEMENT		CORRESPONDING
NUMBER	IND	RSSPONSE QUANT.
1	1	STRESS(MESSEBENE A)
7	2	STRESSIMESSEBENE CII
15	3	STRESSIMESSEBENE M11
17		STRESSIMESSEBENE E)

21	5	STRESSIMESSEBENE	FI
EXTERNAL		FORCE DATA	
NODE		CORRESPONDING	
NUMBER	IND	RESPONSE QUANT.	
11	1	FORCE	
18	2	FORCE	
37	3	FORCE	
43		FORCE	
49	5	FORCE	

* NOTE- THESE ARE THE EASE2 NODE AND ELEMENT NUMBERS(FEB 81)

EZHIST D A T A

VARIABLES

TIME	- TIME AT WHICH HISTORY RESULTS ARE SAVE: ON THE	
×1(J)	• VALUE OF THE J-TH REQUEST QUANTITY AT SOLUTION TIME - TIME	
. SET I	. L (INTEGER TIME . 1). THIS IS FOR LOOP FROM STATMENT 10 TO 20.	

10 1 = 1

00000

C 00 20 ILOOP=1, 10T IFINODII, ITPHI.NE.OI GOTO 20 READIE2HIST) TIME, (XI(J), J=1, MAXCOL) IFIISKIP.EQ.1) GOTO 60 C ** GENERATE ARRAY OF SOLUTION TIMES TTIMEIII . TIME #RITE(TTAPE) TTIME(1) 64 CONTINUE C ** BRANCHING DEPENDENT UPON TYPE OF DATA IF(KEY.EQ.1) GOTO 55 IFINEY.EG.21 GOTO 57 IFIKET.EQ.71 GOTO 56 .. NODE DATA FOR OUTPUT C 55 CONTINUE IFIL.EQ.421 GOTO 70

C . DISPLACEMENT DATAI IN THE FOLLOWING D . TO .XI . WHERE D ARE THE LOCAL

C-6

DISPLACEMENTS, TD IS THE TRANSFORMATION MATRIX, AND XI ARE THE GLOBAL DISPLACEMENTS, I SA(J.I) = STPB(J.I) + SPA(J.I) SV(J.I) = SGAT(SA(J.I)+02 + SPT(J.I)+2 - SA(J.I)+52T(J.I) + 35708(J.I)+02210+2021(J.I)+21 (STP9+STFN+58EN+5702+25708(J.I)+104211E + BENDING+ TEVSILE+ BENDING+ AND TORSIONAL STRESSES+ RESPECTIVELY+ UNITS ARE -ISPA AND SPT ARE THE AXIAL AND TANGENTIAL STRESS DUE TO INTERNAL PRESSUREIP).RESPECTIVELY. THEY ARE AT INTEAUL NODE POINTS(IND) 3.4.AND 5. UNITS OF STRESS ARE N/M4002 FFMAXCOL.LE.21 G0T0 75
STPRIND.11 = XI[1/1.0E6
STPRIND.1] = XI[5/11.0E6
STPRIND.1] = XI[5/11.0E6
STPRIND.1] = XI[5/11.0E6
STPRIND.1] = XI[5/11.0E6
* PERFORM CALC. \$ MITH EASE2 DATA AND PRESSURE[P] DATA DX(IND.1) * (TD11,1,1ND1*X1(1) * TD(1,2,1ND)*X1(2) * TD(1,3,1ND)*X1(3))*1000.0 0 TV(1N0.1) * (TD12,1,1ND)*X1(1) * TD(2,2,1ND)*X1(2) * TD(2,3,1ND)*X1(3))*1000.0 DZ(1ND.1) * (TD13,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,3,1ND)*X1(3),1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,3,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(3,1,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(2,1,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1) * TD(3,2,1ND)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(1)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(2)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(2)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(2)*X1(2) * TD(3,2,1ND)*X1(3,1,1ND)*X1(2)*X1(2)*X1(2) * TD(3,2,1ND)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2) * TD(3,2,1ND)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X1(2)*X * TA(3,3)*XI(3)1/9.81 (ACCELERATION AT IND IS(AX*AY*AZ)* UNITS ARE G-S*) ATAPE * 13 * IND ISY IS THE COMPARATIVE STRESS. UNITS ARE NIMMORZ.I ÷ . * GOTO 20 CONTENUE • ACCELETION DATAE A • TA•XI 1 AXEND.19 • ETA.119×XIE19 • TAEL.219×EE23 FAEL.319×XEE311/9,081 TA(2+2)*X1(2) TA(3,2)*X[[2] WRITE(ATAPE: AXIIND,11,4X1IND,11,42(IND,1) GOTO 20 ** PIPE DATA FOR DUTPUT GOTO 656 PI2*IP11 = PI2*IP11 - PI2*I1 SPA(5*I1 = 0.100P12*IP11/R2M1(2) SPT(5*I1 = 285PA(5*I1 P15.[P1] * P15.[P1] ~ P15.[] SP4(3.[] * 0.10*P15.[P11/R2M115) SP1(3.[] * 2*SP4(3.[] PI4+[PL] * P[4+[PL] ~ P[4+L]
SPA[4+L] * 0.100P[4+[PL]/R2M[[4]
SPT[4+L] * 205Pa[4+L] GO TO (681,682,683,684,6851,IND . ÷ # (TA(2,1)*XI(1) A2([N0.[] = [TA[3.1]*X[[]] # TALZ . 31 * XI [ 311 /9.81 SPAIL.11 = 0.0 SPTIL.11 = 0.0 SP4(2+11 = 0.0 SPT12+11 = 0.0 1 + 1100P + 1 * 2 * * # # / N AYEIND.IJ 6010 686 GOTO 686 CONTINUE G010 686 CONTINUE 0N1 = 1 . . 01 685 680 36 683 489 631 682 0000 U. 0 000 00 U. 0 1

C-7

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FORMATIINI, 43HOUTPUT CALC. RESULTSIOUTPUT EVERY ** STEPSI,/////
WRITEI0.1311
FORMATIIX.10HTIME ARRAY.//)
K = 0
                                                                                                                                                                                                                                                                                                                                                                                                                                                               INV AND ME ARE THE Y AND 2 BENDING MOMENTS, UNITS ARE NON.)
IALPHA IS THE ANGLE OF THE BENDING AXIS, UNITS ARE DEGREES.)
ALTAPE - 19 + IND
Stare = 14 + 1ND
WRITE(STAPE) STENLIND,11,58ENLIND,11,57CK(IND,11,5PT(IND,11,
* SV(IND,1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                *
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        .
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      FX(IND.1) = TF(1,1,1ND)=XI(1) + TF(1,2,1ND)=XI(2)
= TF(1,3,1ND)=XI(3)
= TF(1,3,1ND)=XI(3)
= TF(2,3,1ND)=XI(3) + TF(2,2,1ND)=XI(2)
= TF(2,3,1ND)=XI(3)
= TF(2,3,1ND)=XI(3)
= TF(3,3,1ND)=XI(3)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           ** TRANSFER TO -5- TO READ NEXT HEADER RECORD
GOTO 5
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             (FCACE AT IND IS (FX,FT,FL), UNITS ARE N.)
FTAPE * 24 + IND
WRITE(FTAPE) FX(IND,1),FT(IND,1),FL(IND,1)
                                                                                                                                                                                                                      MZZS - MZEJ.II
IFEMY*LT.ZEBOMDI GOTO 929
ALPMAEJ.II - ATANZEMZEJ.II.MYEJ.II1+57.3
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   · FOR OUTPUT TO PRINTER VIA TAPES I - 1
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                    ** FORCE DATA FOR OUTPUT! F * TF#KI)
                                                                                                                                                                                                                                                                                                                        IFINZZ-LT-ZEROMO) GOTO 928
IFINZZS-LT-0.01 ALPHAIJ-I1 ~ 90.0
IFINZZS-LT-0.01 ALPHAIJ-I1 ~ -90.0
GOTO 927
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           . IN THIS LOOPILO TO 201 L - 1 .
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          IALL ITENS HAVE BEEN CALCULATED. I
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          * REWIND TAPES 10 THRU 29
D0 579 J=10.29
REWIND J
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               WRITEIALTAPES ALPHAIIND.18
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             .. welte out to output ..
                                                         G0T0 20
CONTINUE
TYCIN0+1) + XI(1)
*/(1/00+1) + XI(2)
                                                                                                                                                                                  ATT - ABSEMYLJ. TT
                                                                                                                                                                                                      $11" FIZHISBY - 274
                                                                                                                                                                                                                                                                                                                                                                                                                             ALPHAIJ.II * 0.0
COVIINUE
                                                                                                                                                               ZEROND = 0.001
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           #RITE16, 3071
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       ISKIP . 1
                                                                                                                                                                                                                                                                                    6010 927
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        CONTINUE
                                                                                                                                                                                                                                                                                                      CONTINUE
                                                                                                                                                                                                                                                                                                                                                                                                       CONTINUE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   CONTINUE
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  CONTINUE
                                                                                                                                           0N7 *
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                  6010 20
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                           1 - 1
                                                                               52
                                                                                                                                                                                                                                                                                                      929
                                                                                                                                                                                                                                                                                                                                                                                                         926
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                                                                                                                                                                                                                                                                                                                                                                                                                                               126
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STEM(IND+I)+S8Ew(IND+I)+ALPHA(IND+I)+STDR(IND+I)+ SFT(IND+I)+S4(IND+I) 944 48116(6.314) 314 FO2MAFT[HL, 8H5TRE55E5+//) 00 315 [ND-1,5 WRITE(6.316) IND WRITE(6.316) IND 316 FO2MATT[H, 60HIND - 15,/,8X,4H5TEN,14X,4H5BEN,13X,54ALP4A,14X, x = 0 `
STAPE = 14 + IND
ALTAPE = 19 + IND
ALTAPE = 19 + IND
00 643 J=1:10T
20 643 J=1:10T
20 643 J=1:10T
20 643 J=1:10T FORMATILH , 5MIND . , 15, /, 8X, 2MDX, 14X, 2MDY, 14X, 2402./1 FORMATILH , 7HIND - 1./ . 8X. 2HAX. 14X. 2HAY. 14X. 2HAZ. / ) WRITE(6,311) DX(INO,1),0Y(INO,1),0Z(IND,1) COMTINUE 00 642 J-1,107 READ(ATAPE: AX(IND,1), AY(IND,1), AZ(IND,1) DTAPE = 10 + 1ND D0 641 J=1.10T &EADIOTAPE: DX(IND+1),DY(IND+1),DZ(IND+1) WRITE(5,311) AX(1,1),AY(1,1),AZ(1,1) COMTINUE #RITE16.312) FORMATFIN1.13MACCELERATIONS.//1 WRITE16.3131 F0%MAT(1X,13H015PLACEMENTS,//) D0 310 1N0-1,3 MRITE(5,309) 1ND READIALTAPET ALPUALIND.11 3(E12.4,4X1) ARITE(6, 332) TTIME(1) K * K * 1 IFIK.EQ.M21 COTO 612 GOTO 642 IFIX. CO.M21 GOTO 613 IFIK.E0.M81 G010 610 G010 600 00 500 J=1,107 READ(TTAPE) TTIME(1) FIX.EQ.M21 60T0 611 FORMAT(3X, 8E15.4) #RITE(6, 3331 ATAPE = 13 . IND #RITE(6.317) #RITE (6, 308) FORMATILHLI + SVIINO.II K * K * 1 * * * * 6010 543 443 CONTINUE 6010 641 SUNTINUS. CONTINUE DRMAT( K + K 613 K + 0 0 = ¥ 0 · · . . . 0 . . ONI . 249 612 332 113 308 611 310 312 113 610 309

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IDate = 6H810327
Call DOSYSMITTIME.TIMSTP.NFRAN.IDT.XOUT.42.ICHAN.IN4.IDATE.01
* Set I = 1 FOR GEN. GERMAN DUTPUT.
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                     STAPE * 14 * INO
READISTAPE: STEWIIND,1:,SBEWIIND,1:,STORIIND,1:,SPTIIND,1:,
                                                                                                                  F0%MATTLHL, 0HF0RCES,//1
00 321 IND*1,5
WRIFE16.3221 IND
F0@MATTLH $6MIND = $15*/*8%*2HF%$14%;2HFY*14%*E4FL;//
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       WRITE(6,311) FX(INO,1),FY(INO,2),F2(INO,1)
CONTINUE
CONTINUE
                                                                                                                                                                                                                                                                                                                              FT&PE = 24 + 1ND
D0 644 J=1,1DT
READIFTAPE1 FX(INO,1),FY(INO,1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   ** DUTPUT DATA TO TAPE USING DUSYS FORMAT
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              ATAPE = 13 + 1ND
READIATAPE! AKIIND,11,4YIIND,11,AZIIND,11
00 401 IND=1,5
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        DTAPE = 10 + IND
READ(DTAPE: DX(IND+1)*OY(IND+1)*DZ(IND+1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                        * GENERATE DATA FOR FILEI (DOSYS FORMAI)
00 696 J*10-29
Rewind J
Timstp 0.0002
NTRAN = *2
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                       00 402 IND=1.5
ALTAPE - 19 + IND
ALTAPE - 19 + IND
ALTAPE - 19 + IND,11
TIME(1) + TITME(1) - 0.0966
XOUT(1, 1) - 01(1,1)
YOUT(1, 2) - 01(1,1)
CONTINUE
FCRMATI 61612.4.6X11
MRITE16.3201
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              X 0UT(1,12) = A2(1,1)
X00T(1,13) = STEN(1,1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                   X0UTII: 3) = 0.211.
X0UTII: 41 = 0.212.
X0UTII: 51 = 0.212.
1.0211: 51 = 0.212.
1.0211: 51 = 0.212.
X0UTII: 61 = 0.213.
X0UTII: 70 = 0.213.
X0UTII
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             IFIK.EQ.M21 60T0 614
60T0 644
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                              00 500 IL00P = 1,IDT
#E&DITTAPE) TTIME(1)
20 400 IND=1,3
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                ICHAN = 30
00 833 J=1.42
IMHIJ1 = J
                                                                                                                                                                                                                                                                                                                                                                                                                                                            K * K * 1
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      * SV(IND+1)
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                               x0uffl. 2
x0uffl. 3
x0uffl. 4
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      I = 0N1
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                      N = 0
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                             1 * 1
                                                                                                                                                                                                                                                                                                     a 10
315
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XOUT(1.14) = SBEN(1.1)
    XOUT(1.15) = ALPHA(1.1)
    XOUT(1.16) = STOR(1.1)
    XOUT(1.17) = 0.0
    XOUT(1.18) = 0.0
    XOUT(1.19) . STEN(2.1)
    XOUT(1.20) - SBEN(2.1)
    XOUT(1,21) - ALPHA(2,1)
     XOUT(1,22) = STOR(2,1)
    X001(1.23) = 0.6
    XOUT(1.24) = 3.0
     XOUT(1,25) = STEN(3,1)
    XOUT(1.26) = 58EN(3.1)
    XOUT(1.27) = ALPHA(3,1)
     XOUT(1,28) = STOR(3,1)
     XOUT(1,291 . SPT(3,1)
     XOUT(1.30) * SV(3.1)
    XOUTII,311 = STEN(4.1)
     XOUT(1,32) = SBEN(4,1)
     XOUT(1.33) = ALPHA(4.1)
    XOUT11.341 - STOR14.11
     XOUT(1.35) = SPT(4.1)
     XOUTI1.361 - SV(4.1)
     XOUT(1,37) = STEN(5,1)
     XOUT(1,38) = 58EN(5.1)
     XOUT(1.39) = ALPHA(5,1)
     XOUT(1,40) = STOR(5,1)
     XOUT(1.41) = SPT(5.1)
     XOUT(1.42) = SV(5.1)
     CALL DOSYSWITTIME, TIMSTP, NTRAN, 1, KOUT, 42, ICHAN, IMH, IDATE, 1)
500 CONTINUE
     . GENERATE DATA FOR FILEZ (DOSYS FORMAT)
     REWIND 10
     VIRAN # 15
     [CHAN = 31
    CALL DOSYSWITTIME, TIMSTP,NTRAN, IDT, XOUT, 42, ICHAN, INH, IDATE, 01
    . SET I . I FOR DUTPUT OF FORCES.
     I = 1
     00 595 ILOOP=1,IDT
     READITTAPE: TTIME(1)
     00 530 IND=1.5
     FTAPE = 24 + IND
530 READIFTAPE: FX(IND,1), FY(IND,1), FZ(IND,1)
     TTIME(1) = TTIME(1) - 0.0966
     XOUT(1. 1) = FX(1.1)
     XOUT(1. 2) = FY(1.1)
     XOUT(1, 3) - FZ(1.1)
     XOUTIL, A) . FX12.11
     XOUT(1, 5) . FY(2,1)
     XOUT(1. 6) = FZ(2.1)
     XOUTI1. 71 - FX(3,1)
     XOUT(1, A) = FY(3.1)
     XOUT(1, 9) - F2(3,1)
     XOUT[1.10] = FX(+.1)
     XOUT(1.11) . FY(4.1)
     XOUT(1,12) . FZ(4,1)
     XOUT[1.13] = FX(5.1]
     XOUTII.14) . FY(5,1)
     XOUT(1.15) = F2(5.1)
     CALL DOSYSWITTIME, TIMSTP, NTRAN, 1, XOUT, 42, ICHAN, INH. IDATE, 1)
595 CONTINUE
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C-11

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CC

C
ENDFILE 30 ENDFILE 30 ENDFILE 31 STOP ENDFILE 31 ENDFILE 30 ENDFILE 30

```
SUBROUTINE DOSYSHISTRTIN, TIMSTP, NTRAN, NPTS, TDATA, MAKTRN, ICHAN,
    .
                   INHERE . IDATE . IFLAG!
0
PURPOSE THIS ROUTINE IS USED TO DUTPUT DATA TO JOSYS FORMAT
            TAPES.
            THE FORMAT STATEMENTS IN THIS ROUTINE ARE SPECIFIC TO
     NOTE
            THE NSF PROJECT 1182.08.
     INPUTS STRTIM START TIME OF THE DATA
            TIMSTP DELTA TIME FOR EACH STEP
            NTRAN NUMBER OF TRANSDUCERSICHANNELS OF DATA1
            NPTS NUMBER OF TIME POINTS
            TOATA TRANSDUCER DATA
            MAXIRN MAXIMUM TRANSOUCERS, USED IN DIMENSIONING TOATA
            ICHAN CHANNEL NUMBER OF DOSYS TAPE
            IWHERE INDICATES WHICH SETS OF TOATA ARE USED AS WHICH
                  TRANSDUCERS. A O INDICATES THAT ZERDES ARE TO
                  BE OUTPUT FOR THAT CHANNEL.
            IDATE INTEGER DATEIL.E. FEB 3,1681 = 810203
            IFLAG IF.LT.O OUTPUT HEADER AND DATA
                  IF.EQ./ OUTPUT HEADER ONLY
                  IF.GT./ DUTPUT DATA ONLY
     OUTPUTS ALL OUTPUT IS TO THE DUSYS TAPE OR THE LISTING FILE(6).
MGD
            DATE
                 BY
                          REASON
     1.0
            3/81
                  LJS
                          URIGINAL
DIMENSION TDATAINPTS, MAXTRN), IWHERE(NTRAN), OUTDAT(256)
     CHECK FOR WHETHER TO OUTPUT HEADER OF NOT
C
C
      IFIIFLAG1 50,50,210
  50 CONTINUE
C
      WRITE OUT HEADER
C
C
      WRITE/ICHAN, 90001 IDATE, IDATE, NTRAN, NPTS
     IFENTRAN.NE.42) GOTO 100
C
      IF 42 CHANNELS, MUST BE THE TRANSDUCER INFORMATION
C
     WRITE(ICHAN, 9001)
     WRITE(ICHAN, 9101)
     #RITELICHAN, 9201)
     WRITE([CHAN, 9301]
      HRITELICHAN, 9401)
     WRITEIICHAN, 9501)
      #RITELICHAN, 96011
      GOTO 200
C
C
      IF OTHER THAN 42, MUST BE FORCES.
C
 100 CONTINUE
      N=NTRAN
```

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```
IF(N.GT.9) N=9
       00 120 1+1.4
         #RITE(1CHAN. 90021 1
  120 CONTINUE
       IFINTRAN. 1.101 GOTO 200
       N = NTRAN
       [FIN.GT.99] N = 99
       DO 130 I = 10.N
         WRITE(ICHAN, 9003) 1
  130 CONTINUE
       IF INTRAN.LT.1001 GO TO 200
       00 140 I - 100,NTRAN
         WRITELICHAN, 9004) I
  14C CONTINUE
  200 CONTINUE
       IF(IFLAG.EQ.O) RETURN
C
C
       OUTPUT DATA POINTS
C
  210 CONTINUE
       TIME . STRTIM
       00 400 I = 1.NPTS
         00 300 J = 1.NTRAN
           IFIINHERELJI.EQ.0) COTO 250
           OUTDAT(J) = TDATA(IWHERE(J),I)
           COTO 300
  250
           CONTINUE
           OUTDATIJI = 0.0
  300
         CONTINUE
         WRITELICHAN, 90051 TIME, LOUTDATLJI, J=1,NTRANI
  400 CONTINUE
       RETURN
. С
C
       FORMATS
C
 9000 FORMATIA6/6HEV 3000./5HV60.4./4HCANC./A6./.
     .
             13H5P4AHDRSR¥350////////////5/15)
 9001 FORMATE 10HR52201 MM.19X.3HL02.10X.6H003400./
              17HHEG IN Y-RICHTUNG ./
               10HR52202 MM. 19X, 3HL02, 10X, 6H003400,/
              17HWEG IN X-RICHTUNG ./
               10HR 52203 MM, 19X, 3HL 02, 10X, 6H003400,/
              17HJEG IN Z-RICHTUNG,/
               10HR52204 MM, 19X, 3HL02, 10X, 6H003400,/
              17HWEG IN Y-RICHTUNG ./
               10HR 52 205 M4, 19X, 3HL 02, 10X, 6H003400,/
              17HWEG IN Z-RICHTUNG ,/
               10HR52206 MM.19X.3HL02.10X.6H003400./
              17 VEG IN X-RICHTUNG)
 9101 FORMATE . H554004 MM. 19X. 3HARS. 10X. 6H000000./
              17HWEG IN Z-RICHTUNG ./
               10H554005 MM. 19X. 3HARS. 10X. 6H000000./
               17HWEG IN X-RICHTUNG ./
               10H554006 MM. 19X. 3HARS. 10X. 6H000000./
               17HWEG IN Y-RICHTUNG,/
               10HSS4001 G +19X+3HARS+10X+6H -250+/
               28HBESCHLEUNIGUNG IN X-RICHTUNG ./
               10HSS4002 G .19X, 3MARS, 10X, 6H -250,/
               28H8ESCHLEUNIGUNG IN Y-RICHTUNG ./
               10H554003 C ,19X, 3HARS, 10X, 6H -250,/
               28HBESCHLEUNIGUNG IN Z-RICHTUNG)
  9201 FORMATE 13HRK1010 N/MM2+/
               11HZUGSPANNUNG ./
      .
```

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.

```
13HRK2010 N/MM2./
             13HBIEGESPANNUNG ./
              12HRK2210 GRAD ./
             21HWINKEL DER BIEGEACHSE ./
              13HRK3010 N/MM2./
             16HTORS IONS SPANNUNG . /
              13HRK4110 N/MM2+/
             35HTANGENTIALSPANNUNG DURCH INNENDRUCK ./
              13HRK5010 N/MM2,/
             18HVERGLEICHSSPANNUNG)
 9301 FORMATE 13HRK1011 N/MM2./
             11HZUGSPANNUNG ./
              13HRK2011 N/MMZ ./
             13HBIEGESPANNUNG ./
              12HRK2211 GRAD +/
             21HWINKEL DER BIEGEACHSE+/
              13HRK3011 N/MM2./
             16HTORSIONSSPANNUNG ./
              13HRK4111 N/MMZ./
             35HTANGENTIALSPANNUNG DURCH INNENDRUCK ./
              13HRKSOIL N/MM2./
             18HVERGLEICHSSPANNUNG)
 9401 FORMATE 13HRK1012 N/MM2./
             11HZUGSPANNUNG ./
              13HRK2012 N/MM2 ./
             13HBIEGESPANNUNG ./
              12HRK2212 GRAD ./
             ZIHWINKEL DER BIEGEACHSE./
              13HRK3012 N/MM2./
             16HTORSIONSSPANNUNG ./
0
              13M#K4112 N/MM2./
15
             35HTANGENTIALSPANNUNG DURCH INNENDRUCK ./
              13HRK5012 N/MM2./
             18HVERGLEICHSSPANNUNG)
 9501 FORMATE 13HRK1013 N/MM2./
             11HZUGSPANNUNG ./
              13HRK2013 N/MM2 ./
             13HBIEGESPANNUNG ./
              13HRK2213 N/MM2./
             21HWINKEL DER BIEGEACHSE./
              13HRK3013 N/MM2./
             16HTORSIONSSPANNUNG ./
              13H2K4113 N/MM2./
             35HTANGENTIALSPANNUNG DURCH INNENDRUCK ./
              13HRK5013 N/MM2,/
             18HVERGLEICHSSPANNUNGI
 9601 FORMATE 13HRK1014 N/MM2./
             11HZUGSPANNUNG ./
              13HRK2014 N/MM2./
             13HBIEGESPANNUNG ./
             " 13HRK2214 N/MM2./
             21HWINKEL DER BIEGEACHSE+/
              13HRK3014 N/MMZ ./
             16HTORS IONS SPANNUNG . /
              13HRK4114 N/MM2+/
              35HTANGENTIALSPANNUNG DURCH INNENDRUCK ./
              13HRK5014 N/MM2./
              18HVERGLEICHSSPANNUNG:
 9003 FORMATE 4HRFOD, 12,3H N.19X.3HL02,10X.6H...../1
 9005 FORMAT(6E12.5.8X)
      END
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