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Results of Crack-Arrest Tests on Two Irradiated High-Copper Welds

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Prepared by S. K. Iskander, W. R. Corwin, R. K. Nanstead

Oak Ridge National Laboratory

Prepared for U.S. Nuclear Regulatory Commission

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Results of Crack-Arrest Tests on Two Irradiated High-Copper Welds

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ACRONYMS

- ASME American Society of Mechanical Engineers
- CMOD crack mouth opening displacement
- CVN Charpy V-notch
- dpa displacements per atom
- DX duplex
- EPRI Electric Power Research Institute
- HSSI Heavy-Section Steel Irradiation
- ORNL Oak Ridge National Laboratory
- NDT nil-ductility-transition temperature, as determined by the drop-weight test according to ASTM E 208
- NRC U.S. Nuclear Regulatory Commission
- RPVs reactor pressure vessels

RT_{NDT} reference nil-ductility-transition temperature, determined in accordance with Subarticle NS-2330 of ASME Boiler and Pressure Vessel Code, Sect. III

WE weld-embrittled

SYMBOLS

- a initial slot length ao or final crack length a.
- a, arrested crack length
- a, initial slot length
- B specimen thickness (Fig. 2)
- $B_{\rm N}$ specimen thickness at crack plane (Fig. 2)
- D split-pin hole diameter (Fig. 2)
- δ crack mouth opening displacement

ATT41-J shift in the 41-J CVN-impact energy level

E Young's modulus

f(x) crack-arrest specimen calibration function [Eq. (2)]

- ϕ fluence, neutrons/cm²
- 2H specimen height (Fig. 2)
- K, value of the stress intensity factor shortly after arrest"

 K_o value of the stress intensity factor at crack initiation

K₁ stress intensity factor

 $K_{1\alpha}$ value of the crack-arrest fracture toughness K_α for a crack that arrests under conditions of crack front plane-strain *

- K1e plane-strain fracture toughness
- K_{Je} a measure of fracture toughness calculated from the J-integral J_e at the point of cleavage using the relationship $(K_{Je})^2 = EJ_e$
- K_o value of the stress intensity factor at crack initiation L.L' (See Fig. 2)

*Excerpted from ASTM E 1221-88.

SYMBOLS

- N slot width

- T test temperature W nominal width of a crack-arrest specimen W_t total width of a crack-arrest specimen (Fig. 2) x fractional crack depth a/W

FOREWORD

The work reported here was performed at Oak Ridge National Laboratory (ORNL) under the Heavy-Section Steel Irradiation (HSSI) Program, W. R. Corwin, Program Manager. The program is sponsored by the Office of Nuclear Regulatory Research of the U.S. Nuclear Regulatory Commission (NRC). The technical monitor for the NRC is A. Taboada.

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ABSTRACT

The objective of this study was to determine the effect of neutron irradiation on the shift and shape of the lowerbound curve to crack-arrest data. Two submerged-arc welds with copper contents of 0.23 and 0.31 wt % were commercially fabricated in 220-mm-thick plate. Crack-arrest specimens fabricated from these welds were irradiated at a nominal temperature of 288°C to an average fluence of 1.9×10^{19} neutrons/cm² (>1 MeV). Evaluation of the results shows that the neutron-irradiation-induced crack-arrest toughness temperature shift is about the same as the Charpy V-notch impact temperature shift at the 41-J energy level. The shape of the lower-bound curves (for the range of test temperatures covered) did not seem to have been altered by irradiation compared to those of the ASME K₁₀ curve.

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1. INTRODUCTION

In the fracture mechanics integrity analysis of reactor pressure vessels (RPVs), the initiation and arrest fracture toughness curves as described in Sect. XI of ASME Boiler and Pressure Vessel Code are often used. These curves are used also for the normal operation of RPVs. The effects of neutron irradiation on toughness are accounted for by shifting the curves upward in temperature without change in shape by an amount equal to the temperature shift of the Charpy V-notch (CVN) impact energy curve at the 41-J level (ΔTT_{k1-J}) . Such a procedure implies that the shifts in the fracture toughness curves are the same as that of the CVN ΔTT_{k1-J} and that irradiation does not change the shapes of the fracture toughness curves.

It is well known that irradiation of some RPV ferritic steels to fluences on the order of 2×10^{19} neutrons/cm² (>1 MeV) can cause changes in the shape of the CVN impact energy curve. To determine whether similar changes in shape can occur in the fracture toughness curves, particularly if such changes could lead to non-conservative determinations of the irradiated fracture toughness, research programs are sponsored by the U.S. Nuclear Regulatory Commission (NRC) within the Heavy-Section Steel Irradiation (HSSI) Program at Oak Ridge National Laboratory (ORNL).

Two of these programs are the Fifth and Sixth Irradiation Series. The objective of the Fifth Series was to determine the effect of neutron irradiation on the shift and shape of the K_{Ic} vs $(T-RT_{NDT})$ curve, where K_{Ic} is the plane-strain fracture toughness, T is the temperature, and RT_{NDT} is the reference nil-ductility-transition temperature. Although the objective is similar, the Sixth Series investigates the effect on K_{Ia} , the plane-strain crack-arrest fracture toughness. Both programs investigate the effects of irradiation on the fracture toughness of welds, since some pressure vessels in operation have welds with copper contents and end-of-life fluences which make them susceptible to severe degradation in toughness. The amount of experimental data on the effects of irradiation on crack-arrest fracture toughness is rather meager (Refs. [1] and [2]).

Two submerged-arc welds with copper contents of 0.23 and 0.31 wt * were commercially fabricated in 220-mm-thick plate. In the Fifth Irradiation Series, irradiated CVN impact, tensile, drop-weight, and compact specimens, made from the weldment were tested and the results are given in Refs. [3] and [4].

Crack-arrest specimens fabricated from these welds were irradiated at a nominal temperature of 288° C to an average fluence of 1.9×10^{19} neutrons/cm² (>1 MeV). This report compares the results of crack-arrest tests on 36 irradiated weld-embrittled-type specimens with those from unirradiated control specimens. Since this is only the first phase of a two-phase program, the conclusions presented here are preliminary.

2. DESCRIPTION OF MATERIALS AND PROCEDURES

The weld wire for both programs was produced commercially in one melt. The melt was split to allow for copper additions and resulted in two weld wires that were comparable in chemical composition except for copper. Several meters of weldment were commercially fabricated from each weld wire and were designated 72W (0.23 wt % Cu) and 73W (0.31 wt % Cu). The welds were commercially fabricated in A 533, grade B, class 2 plate of 220-mm (8 5/8-in.) thickness by using the submerged-arc weld process with one lot of Linde 0124 flux. A macrograph of weld 72W is shown in Fig. 1, and the chemical compositions of both welds are given in Table 1. The welds were given a postweld heat treatment of 607°C for 40 h, typical of that given commercial RPVs.

Two capsules, each containing 30 compact crack-arrest specimens of the two weldments, have been irradiated at a nominal temperature of 288°C to a fluence of approximately 1.9×10^{19} neutrons/cm² (>1 MeV) in the Oak Ridge Research Reactor located at the ORNL. The irradiated specimen complement is given in Table 2, which gives the overall specimen dimensions. The complete nominal specimen dimensions are given in Fig. 2 and Table 3. For administrative purposes, the program was conducted in two phases. This report presents results of Phase 1 which consisted of crack-arrest tests on the 36 weld-embrittled-type specimens. In Phase 2, the results from remaining 24 duplex-type crack-arrest specimens will be reported.

2.1 Testing Procedures

Testing was performed according to the ASTM "Test for Determining Plane-Strain Crack-Arrest Fracture Toughness, K_{Ie} , of Ferritic Steels" (E 1221-88). Crack-arrest testing of the irradiated specimens was performed in a hot cell by using a servohydraulic machine and the test chamber shown schematically in Fig. 3, and photographically in Fig. 4. This equipment allows specimens to be tested with the split pins in either the so-called "normal" or "inverted" configuration at test temperatures ranging from -100 to 260°C.⁺ In the "normal" configuration, the lateral surfaces of the specimen are in contact with the test-machine platen and friction decreases the crack-driving force somewhat compared with the "inverted" configuration in which the specimen sits on the narrow shoulders of the split pin. The "inverted" configuration is used at testing temperatures above NDT. For further information on the "normal" and "inverted" test configuration in crack-arrest testing, see Note 5 of ASTM E 1221-88.

^{*}This temperature range was chosen during the design phase of the test chamber on the basis of the anticipated test temperature. The maximum temperature is limited by the Teflon™ insert in the contact thermocouple.

ORNL-PHOTO 4145-90

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Fig. 1. Cross section of typical submerged-arc weld used in the 72W and 73W welds used in the Fifth and Sixth Irradiation Series. ORNL-PHOTO 4145-96



Fig. 1. Cross section of cypical submerged-arc weld used in the 72W and 73W welds used in the Fifth and Sixth Irradiation Series.

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Material	Composition (wt %)												
	C	Mn	Р	S	Si	Cr	Ni	Mo	Cu	٧			
72W	0.093	1.60	0.006	0.006	0.44	0.27	0.60	0.58	0.23	0.003			
73W	0.098	1.56	0.005	0.005	0.45	0.25	0.60	0.58	0.31	0.003			

Table 1. Chemical composition of the two submerged-arc welds in the Fifth and Sixth Irradiation Series

Source: Values based on formula from R. K. Nanstad et al., "Effects of Irradiation on K_{1c} curves for High-Copper Welds," pp. 214-33 in Effects of Radiation on Materials, 14th International Symposium, ASTM STP 1046, Vol. II, ed. N. H. Packan, R. E. Stoller, and A. S. Kumar, American Society for Testing and Materials, Philadelphia, 1990.

Specimen type	Dimensions (mm)	Quantity per weld		
Weld embrittled	25 × 76 × 76	8		
Weld embrittled	$25 \times 152 \times 152$	7		
Weld embrittled	$33 \times 152 \times 152$	3		
Duplex	$33 \times 152 \times 152$	12		

Table 2. Irradiated crack-arrest specimen complement for each of the 72W and 73W weldments



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Nominal specimen size	u	a _o	a _o /¥	В	u,	211	L	D	Ľ'
			72¥ and 73¥	weldments					
25 × 76 × 76	63.50	20.96	0.33	25.40	76.20	76.20	12.70	21.06	44.4
			72¥ wel	dment.					
25 × 152 × 152		43.82-44.45	0.345-0.350					84.4	6-85.09
33 × 152 × 152		43.82-45.09	0.345-0.355					83.6	12-85.09
			739 we	dment					
25 × 152 × 152	127.00	49.53	0.39	25.40	152.40	152.40	25.40	31.75	79.38
33 × 152 × 152	127.00	49.53	0.39	33.02	152.40	152.40	25.40	31.75	79.38

Table 3. Nominal dimensions of weld-embrittled crack-arrest specimens used in the Sixth Irradiation Series

(All dimensions are in millimeters)

^aTo accommodate the crack-starter notch within the brittle weld. L' had to be adjusted. The values shown reflect the range for the a_0 , a_0/W , and L' dimensions. All other dimensions are the same for both the 72W and 73W weldments.

NOTE:

- ao initial slot length.
- B specimen thickness.
- D split-pin hole diameter.
- 2H specimen height.
- L,L' = see fig. 2.
 - w nominal width of crack-arrest specimen.
 - W, total width of crack-arrest specimen.



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Fig. 3. Schematic drawing of equipment used to perform crack-arrest testing of irradiated specimens before set-up in the hot-cell 500-kN testing machine.



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Fig. 4. Photograph of equipment used to perform crack-arrest testing of irradiated specimens before set-up in the hot-cell 500-kN testing machine.

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Figure 5 shows a closeup view of the environmental chamber used for temperature conditioning of the crack-arrest specimens after it was installed in the hot cell 500-kN tensile testing machine. Figure 6 gives a general view of the test system as it is being prepared for a crackarrest test.

Temperature conditioning below ambient is obtained by venting cold nitrogen gas into the chamber containing the crack-arrest specimen. Above ambient, temperature control is achieved by means of six heater elements. Four of these elements are cartridge type and are embedded in the base of the fixture. Two U-shaped heater elements can be moved into place below or above the specimen. After the specimen is positioned in the fixture, a contact thermocouple is lowered onto the specimen surface near the midpoint of the crack path to measure the specimen temperature.

Considerable effort was made to ensure temperature uniformity and accuracy in each crack-arrest specimen. Specimens of A 533B steel and of the same three sizes as those to be tested were instrumented with thermocouples and thermally conditioned in the fixture over the relevant temperature range in both normal and inverted configurations. The uniformity and accuracy of the temperature throughout the specimen were within ±1°C when compared to the values measured by the contact thermocouple used during testing.

The clip gage used for the irradiated specimen tests was specially designed and fabricated at ORNL. Long clip-gage arms were incorporated in the design to measure the crack mouth opening displacement (CMOD) from outside the temperature conditioning chamber. The arms are instrumented with temperature-compensated electrical resistance strain gages. Moreover, tests outside the hot cell have shown that the clip-gage temperature is < 35°C when the specimen temperature is 350°C. Since the maximum test temperature for this series of tests is under 150°C, the clip-gage is calibrated at room temperature with no significant loss in accurscy. The clip-gage was calibrated in the hot cell with a Boekler micrometer before and after every test since the CMOD (not the load) measured during the test is central to the determination of the crack-arrest toughness $K_{\rm B}$. The error in $K_{\rm B}$ due to errors in the measurement of CMOD is estimated to be less than 2%.

An irradiated specimen to be tested is placed in the conditioning chamber, and special "knives" are used to center the specimen beneath the loading wedge. The sides of the loading wedge were covered with a replaceable strip of Teflon" to reduce the friction between the wedge and the split pins. The testing machine ram was then raised until the loading wedge just touches the split pins (a load of about 40 N); then the ram is lowered until no-load is indicated. The centering "knives" were then removed since the proper position of the specimen is maintained by the inserted load wedge. The clip-gage is then seated in conical grooves of gage blocks welded (prior to irradiation) to the crack-arrest

^{*}For the purposes of this test, thermocouples were tack-welded to the test specimen.



Fig. 5. Closeup view of the environmental chamber used for temperature conditioning of the crack-arrest specimens after installation in the hot-cell 500-kN tensile testing machine.



Fig. 6. General view of the test system as it is being prepared for a crack-arrest test.

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specimen. The load vs CMOD is recorded on an X-Y plotter together with the clip-gage calibration marks; a typical chart from a test is shown in Fig. 7. The charts from all the irradiated tests are included in Appendix E. As mentioned above, the value of the load is not used in the calculation of K_a . However, the plot of load vs CMOD is indispensable in the conduct of the test and determination of the CMOD at various points in the loading-unloading cyrles (see ASTM E 1221-88).

2.2 Analysis Procedures

The «rrested crack front is marked by heat-tinting the specimen after testing. The specimen is then cooled with liquid nitrogen and broken open. The length of the remaining ligament for each specimen is obtained by measuring the position of the arrested crack front by using the averaging procedure prescribed in ASTM E 1221-88. In the case of irradiated specimens, a digitizing tablet was used to measure the length of the remaining ligament on a photograph taken through a Kollmorgan" periscope of the broken specimen half. Typical fracture surfaces for two 33. × 152.mm specimens are shown in Fig. 8 (photographs of the fracture surfaces of all the specimens tested are given in Appendix E), Numerous unbroken ligaments, typical for pressure vessel steels at these test temperatures, can be observed on the fracture surface of the specimens. The known dimensions of the specimen serve as the scale for chese measurements. In the case of the unirradiated specimens, the length of the remaining ligament is directly measured by a digital measuring microscope. The errors due to the measurement method are estimated to be less than those due to the shape of the crack front.

To estimate accuracy of the crack-arrest toughness, two values of the tress intensity factor (K_o and K_a) are calculated. The former is the stress intensity factor at crack initiation, and the latter is the value shortly after arrest. Both values are calculated by substituting appropriate values in the following expressions given in ASTM E 1221-88:

$$\kappa = E \delta f(\mathbf{x}) \sqrt{\left[\frac{B}{(B_{\mathbf{x}}W)} \right]}, \qquad (1)$$

where f(x) = crack-arrest specimen calibration function defined as follows:

$$f(x) = \frac{2.24 (1.72 - 0.9x + x^2) \sqrt{(1 - x)}}{(9.85 - 0.17x + 11x^2)},$$
(2)

and

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x = a/W,

E = Young's modulus,

a = initial slot length a, or arrested (final) crack length a,

W = specimen width,

B = specimen thickness,

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Fig. 7. Chart produced on an X-Y plotter during a typical crackerrest test of an irradiated specimen.

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them open. The 2- or 3-mm, darker, transverse strips near the initial crack front surfaces of two typical crack-arrest specimens after heat-tinting and splitting Fig. 8. Photograph taken through a Kollmorgen periscope of the fracture are the brittle crack-starter weld beads. B_N = specimen thickness at crack plane,

δ = CMOD.

If all other values are unchanged the error in K is directly pro_{r} ortional to the error in δ (given above). The error in K due to errors in crack length is estimated as described below.

In Eq. (1), if all other values are constant, the error in K is equal to that of f(x) due to the error in the fractional crack length x. Table 4 shows various estimates of the errors in f(x) resulting from the measurement of the initial slot length a_0 and arrested crack length a_a and for the various specimen sizes. A plot of f(x) vs fractional crack length x, rhown in Fig. 9, is nearly linear in x. However, because f(x) decreases as x increases, the absolute value of the average relative error in f(x) increases as x increases. The average relative error is defined as:

$$\frac{f[(1+e) x] - f[(1-e) x]}{f(x)}$$

Two relevant values of x are 0.33 for the initial slot length and 0.85 for the final crack length. The former is the minimum value for the crack-arrest specimens manufactured for the Sixth Series program, and the latter is the maximum valid value allowed by ASTM E 1221-88. A ±1% change for x = 0.33 results in a ± 0.5 % change in f(x), while the same +1% change for x = 0.85 results in an approxima y ± 3 % change in f(x).

The determination of a_0 is performe with a digital measuring microscope, and the error is estimated to e < 0.2 mm, irrespective of the specimen size. Hence the maximum error in f(x) (and K_0) for the initial slot length is ± 0.5 % for a small specimen and even less for a large specimen. In both cases, the initial slot length is used to estimate K_0 , which gives an estimate of K at the onset of rapid fracture. It is an estimate at best because the crack at initiation is not a sharp one and has a finite root radius. Moreover, K_0 is of interest to the experimenter only.

As mentioned, the measurement of the final crack length a_a is performed according to ASTM E 1221-88 by an averaging procedure. Since it is rare that the arrested crack front is square or straight (Fig. 8), it is difficult to estimate the error in such a determination. From the expression for K_a given in ASTM E 1221-88, estimates of the error were made by using different value, of the arrested crack length to obtain bounds on the error. A maximum value of a_a is 54 mm for a small 25 x 76 x 76 mm specimen since this is the maximum length permitted in E 1221-88 for this specimen size. Errors of ±1 or ±2 mm in a_a for this case result in errors of ± 6.5 % or ± 13 %, respectively, in K_a . A similar procedure was followed for the large (33 × 152 × 152 mm) specimen. The error estimates are summarized in Table 4. The total maximum error in K_a from all sources is estimated to be approximately ±10% for the small specimens and ±5% for the large ones.

		Inita	al cra	ck leng	th a _o	Fit	nal crack	length a,			
			Erro	cs in		Errors in					
Specimen size ^b	W (mm)	a _o (mm)	a _o (mm)	a _o /W (%)	f(x) (%)	a _a (~19)	a _a (mm)	a _a /W (%)	f(x) (%)		
Small	63.5	21	0.2	±1	Ŧ0.5	54	±1 ±2	±2 ±4	∓6.5 ∓13		
Large	127	44	0.2	±0.5	∓0.2	108	±1 ±2	±1 ±2	∓ 3 ∓6.5		

Table 4. Estimate of the average relative error^a in f(x) (and thus in K_a) due to errors in the determination of initial and final crack lengths $x = a_0/W = 0.33$ and $x = a_a/W = 0.85$, respectively, for the various specimen sizes

aDefined as $\frac{1}{2} \left[f[(1+e) \cdot x] - f[(1-e) \cdot x] \right] / f(x)$.

 $b_{\text{The 25} \times 76 \times 76 \text{ mm}}$ specimens are the small specimens, while the 25 (or 33) \times 152 \times 152 mm ones are the large ones.

NOTE:

- a_a = arrested crack length.
- a_o = initial slot length.

f(x) = crack-arrest specimen specimen calibration function.

k_a = value of stress intensity factor shortly after arrest.

W - width.



3. RESULTS AND DISCUSSION

The temperature control for the two irradiation capsules, each containing 30 specimens, was excellent. The temperature most of both weld-embrittled and duplex-type specimens (22 in one capsule and 23 in the other) ranged from 286 to 290°C, while the few remaining specimens were irradiated at temperatures ranging from 280 to 286°C. The average irradiation temperature of the weld-embrittled-type crack-arrest specimens is 286 and 285°C for the 72W and 73W specimens, respectively. For both 72W and 73W specimens, the fluences to which ti, weldembrittled-type crack-arrest specimens were irradiated ranged from approximately 1.4 to 2.4 \times 10¹⁹ neutrons/cm² (>1 MeV), and the average fluence and standard deviation is approximately 1.9 and 0.3 \times 10^{19} neutrons/cm² (>1 MeV), respectively. Bar charts showing the distribution of fluence of the specimens successfully tested, 18 and 17 from the 72W and 73W weldments, respectively, are given in Figs. 10 and 11, respectively. The average irradiation temperature and the exposure values of each specimen are given later in this report. A summary of the irradiation data for the weld-embrittled specimens tested in Phase 1 is given in Table 5. Detailed reports on the dosimetry and operating history of the two capsules will be published separately.

The yield strength and Young's modulus of the test material are required in crack-arrest testing. Young's modulus is calculated from the following expression, from Ref. [5]: E = 207.2 - 0.0571T, where E = Young's modulus in GPa, and T = temperature in °C (discussed also in Appendix C) The unirradiated and irradiated tensile strengths have been reproduced from Ref. [3] in Figs. 12 and 13 for welds 72W and 73W, respectively. The average fluence of the tensile samples, approximately $1.6 \times 10^{\circ}$ neutrons/cm² (>1 MeV), is about 15% less than that for the crack-arrest specimens. This difference will not affect the values of the crack-arrest toughnesses K_a but <u>could</u> have a small effect on the validity of each specimen as it is prescribed in ASTM E 1221-88. The observations in this report are preliminary.

The RT_{NDT} values for both weldments in the unirradiated and irradiated conditions are given in Table 6 (from Refs. [3] and [4]). The initial RT_{NDT} values were determined in accordance with Subarticle NB-2330 of ASME Boiler and Pressure Vessel Code, Sect. III, and are the same as the drop-weight NDT values. The "adjusted" RT_{NDT} values were determined according to paragraph 10.2.2 of ASTM E 185-82 by adding the CVN ΔTT_{41-J} values to the initial RT_{NDT} values. The CVN ΔTT_{41-J} shifts have been determined from a relatively large number of specimens, as shown in Fig. 14 (also from Refs. [3] and [4]). Approximately 84 and 56 unirradiated and irradiated CVN specimens were tested from each of the 72W and 73W welds respectively. The average fluence of CVN-impact energy specimens for both the 72W and 73W weldments is 1.51×10^{19} neutrons/cm²,



Fig. 10. Distribution of fluence in the 18 irradiated weld-embrittled-type crack specimens tested from the 72W weld.



Fig. 11. Distribution of fluence in the 17 irradiated crack specimens of the 73W weld that were tested successfully.

Table 5. Summary of the irradiation temperature and fluence for Series 72W and 73W crack-arrest specimens

	72W (18 s	pecimens)	73W (17 specimens				
	Temperature (*C)	Fluence ^a	Temperature (°C)	Fluence ^a			
Mean	286	1.88	285	1.93			
Standard Deviation Minimum Maximum	281	1.42	280	0.32			

[#]In units of 10^{19} neutrons/cm² (>1 MeV).

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900 UNIRRADIATED 72W WELDS 8 800 \diamond ULT = 630 - 0.8969T + 0.0024T² TENSILE STRENGTH (MPa) CI YIELD = $506 - 0.3808T + 0.0073T^2$ 700 600 500 400 (a) 900 IRRADIATED 72W WELD 800 TENSILE STRENGTH (MPd) 700 D (a) 600 $YIELD = 734 - 0.7314T + 0.0018T^{2}$ 500 $ULT = 636 - 0.6989T + 0.0024T^{2}$ 400 -100 -200 0 100 2.00 (3) TEMPERATURE (°C)



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ORNL-DWG 90-15464 900 UNIRRADIATED 73W WELDS 8 800 \Diamond ULT = 624 - 0.8999T + 0.0021T² TENSILE STRENGTH (MPa) \Box YIELD = 502 - 0.4541T + 0.0061T² 700 600 500 400 (a) 900 IRRADIATED 73W WELD 800 TENSILE STRENGTH (MPa) 700 D 8 600 1 $YIELD = 756 - 0.5656T + 7.126E - 4 T^{2}$ 500 ULT = $661 - 0.5349T + 9.437E - 4T^{2}$ 400 -100 100 0 200 -200 (b) TEMPERATURE (°C)



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Initial RT _{NDT} (°C)		Charpy im observed	pact ^a results	Adjusted ^a	Crack-arre normalized re	Normalized ^b	
		F (10 ¹⁹ n/cm ²) (>1 MeV)	атт ₄₁₋ (К)	(°C)	F' (10 ¹⁹ n/cm ²) (>1 MeV)	ΔTT _{41-J} (K)	RT _{NDT} (°C)
72W	-23	1.51	72	49	1.88	80	57
73W	- 34	1.51	82	48	1.93	93	59

Table 6. Initial, adjusted, and normalized reference temperatures (RT_{RDT}) for the 72W and 73W weldments

^aSource: Data from R. K. Nanstad et al., "Effects of Irradiation on K_{I:} curves for High-Copper Welds," pp. 214-33 in Effects of Radiation on Materials, 14th International Symposium, ASTM STP 1046, Vol. II, ed. N. H. Packan, R. E. Stoller, and A. S. Kumar, American Society for Testing and Materials, Philadelphia, 1990.

^bNormalization: (ΔTT_{41-J})(F'/F)^{0.5}. Source: Adopted with permission from G. R. Odette and G. E. Lucas, "Irradiation Embrittlement of Reactor Pressure Vessel Steels: Mechanisms, Models, and Data Correlations," pp. 206-41 in Radiation Embrittlement of Nuclear Pressure Vessel Steels: An International Review (Second Volume), ASTM STP 909, ed. L. E. Steel, American Society for Testing and Materials, Philadelphia.

NOTE:

- F = fluence for Charpy V-notch impact specimens.
- F' = fluence for crack-arrest specimens.
- n = neutrons.
- ATT_{41-J} = shift in 41-J Charpy V-notch impact energy level.



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while that for the crack-arrest specimens is approximately 1.9×10^{19} neutrons/cm². To normalize the RT_{NDT} shift, an average trend curve applicable to the behavior of test reactor data and having a slope of 0.5 was used (Ref. [6]). It is of the form:

Normalized $\triangle T_{41-J} = (\triangle TT_{41-J}) (\phi' / \phi)^{0.5}$,

where ϕ^{\prime} and ϕ are the average fluences for the crack-arrest and the CVN-impact sp imens, respectively, in neutrons per square centimeter (>1 MeV). Both shifts for each weldmant are given in Table 6. The normalized RT_NDT values were used to shift the ASME K_{Ia} curves shown later in this report.

Tables 7 and 8 give the results of testing the unirradiated crackarrest specimens of the 72W and 73W Series, respectively, and Tables 9 and 10 give data for the irradiated specimens of these two series. These tables also give for each specimen the irradiation temperature and the exposure value [fluence for energies > 1 MeV and > 0.1 MeV as well as displacements per atom (dpa)]. The K_a values have not been normalized to a single average fluence to account for the variation of fluence from one specimen to another.

In testing of one specimen, A73W17 (listed in Table 10), the crack failed to initiate in a cleavage mode and exhibited tearing. It was retested at 30°C, but the crack did not arrest, and the specimen broke. The K_a value fr m specimen A73W37 (tested at -25°C) is recognized as too low because the arrested crack length a_a is 96% of the nominal specimen width W. The expression in ASTM E 1221-88 used to calculate K_a is limited to a_a/W \leq 0.85. While other specimens may be slightly invalid and hence may or may not necessarily represent plane-strain behavior, the authors believe this specimen to be so far beyond the validity limits as to be clearly nonrepresentative of plane-strain crack-arrest toughness of the weld. Hence, its position below the ASME K_{Ia} curve is of no significance. In general, even though some of the other individual specimens may not strictly meet all of the ASTM validity criteria, the authors believe the data to be representative of the arrest toughness that would exist within a thick-walled RPV.

The task group responsible for crack-arrest test procedure ASTM E 1221-88 recognized the need for the data to represent plain-strain conditions. Conservative validity criteria were selected on the basis of existing analytical and experimental results (Ref. [7]) to ensure that specimens meeting those criteria would indeed be representative of planestrain conditions. Even at the time ASTM E 1221-88 was written, however, it was recognized that those criteria might be overly restrictive; they were, nonetheless, chosen to be used until more evidence justifying their relaxation could be amassed. More data are now available; for example, Ref. [8] shows crack-arrest test results clearly violating the size criteria of ASTM E 1221-88, that still accurately predict the arrest behavior of a thick-walled pressure vessel under simulated pressurizedthermal-shock loading. On the bases of these and other considerations, the ASTM task group will be asked to reconsider relaxation of the validity criteria.

Specimen	Test	Ka	Validity ^a
	tempera	ture /	
	(°C)	(MPa•√	m)
N-14		DE - 76 - 76 -	m enerimens
weid-em	prittied	25 X /0 X /0 I	un specimens
A72W37	-61	66	
A72W35	- 60	53	
A72W38	- 59	57	
A72W36	- 47	71	
A72W34	-45	73	
A72W24	- 30	61	
A72W28	- 30	60	
A72W25	- 30	79	В
A72W31	.15	108	E,E
17-14 -mh	deploy 0	E U 150 U 150	mm enecimen
weid-emp	fittied 2	2 × 125 × 126	nun apecimen
A72W44	- 60	76	
A72W39	- 60	45	A
A72W20	- 47	67	
A72W41	-45	74	
A72W46	-45	92	
A72W43	- 32	92	
A72W08	- 30	104	
A72W05	- 30	107	
A72W48	- 30	98	
A72W01	-16	97	
A72W03	-14	128	C
Weld-emb	rittled :	$33 \times 152 \times 152$	mm specimer
A72W45	-45	76	
A72W47	- 30	91	
A72W07	-15	103	
A72W04	-15	54	
A72W19	-15	94	
A72W02	0	93	
A72W12	0	114	
437 AC 11 & St.			

Table 7. Unirradiated crack-arrest toughness

Specimen	Test	Κ _a	Validity ^a
	(°C)	(MPa•√m))
Duple	x 33 $ imes$ 152 $ imes$	152 mm spe	cimens
A72W71	-14	91	
A72W66	- 2	103	
A72W63	- 1	125	
A72W64	1	108	
A72W65	4	125	
A72W68	5	115	
A72W73	5	142	C.D
A72W62	10	136	B,C
A72W57	21	146	C,D
Weld-embr	ittled 51 $ imes$:	203 × 203 m	m specimens
A72W83	+ 30	85	
A72W85	-15	95	
A72W84	0	107	

Table 7. (continued)

⁴One or more letters for a specimen indicate that the results did not meet one of the minimum lengths of the ASTM E 1221-88 validity criteria. The letters correspond to those in Table 2 of ASTM E 1221-88, which can be paraphrased as:

A,B = unbroken ligament too short.

C = specimen too thin.

D,E = insufficient crack-jump length.

NOTE:

 K_a = value of stress intensity factor shortly after arrest.

RT_{NDT} = reference nil-ductility-transition temperature.

	(KINDT -	-34-0)	
Specimen	Test	Ka	Validity ^a
	(°C)	(MPa•√m)	
Weld-embr	ittled 25 \times	76 × 76 mm :	specimens
A73W38	- 62	50	A
A73W42	- 60	58	
A73W32	- 59	56	A
A73W36	- 59	72	
A73W01	- 45	52	
A73W02	-45	75	
A73W03	-45	64	A, B
A73W06	- 35	83	В
A73W04	- 30	67	
A73W05	- 30	70	Α,Β
Weld-emb	rittled 25 \times	152 × 152 m	m specimens
A73W28	-61	69	
A73W43	-45	73	
A73W47	-45	85	
A73W30	- 44	71	
A73W11	- 32	85	
A73W48	- 31	75	
A73W50	- 30	80	
.73W16	-29	89	
A73W52	-29	77	
A73W20	-16	126	C
A73W25	-15	141	B,C
Weld-emb	orittled 33 $ imes$	152 × 152 1	mm specimen
A73W29	- 30	100	
A73W10	-16	77	A
A73W22	-16	110	
A73W46	-15	124	
A73W44	-15	123	
A73W49	-15	117	
A73W24	-13	89	A, B
A73W27	-12	111	

Table 8. Unirradiated crack-arrest toughness data obtained from testing weldment 73W (RT_{EDT} = -34°C)

Specimen	Test temperature (°C)	K _a (MPa•√m)	Validity
Duplex 3	3 × 152 × 152	mm specimens	
A73W86	-5	101	
A73W07	5	129	B
A73W08	5	119	5
A73W09	5	112	
A73W85	5	137	C.D
A73W87	5	113	- 1-
A73W88	15	132	

Table 8. (continued)

Duplex 51 \times 203 \times 203 mm specimen

A73W75	5	107
and the set of the		701

⁴One or more letters for a specimen indicate that the test results did not meet one of the minimum lengths of the ASTM E 1221-88 validity criteria. The letters correspond to those in Table 2 of ASTM E 1221-88, which can be paraphrased as:

MINANTON ALLONDING LOU BIDIC.	A,B	101	unbroken	11	gament	too	short.
-------------------------------	-----	-----	----------	----	--------	-----	--------

C = specimen too thin.

D,E = insufficient crack-jump length.

NOTE:

Ka	-61	value of	f stre	\$S	intensity	factor
		shortly	after	ar	rest.	

RT_{NDT} = reference nil-ductility-transition temperature.

Specimen	Test	K.	Irradiation	Exposure	values	Displacements	Validity ³
	temperature (*C)	erature (MPa•√m) temperature *C) (°C) I		(MPa-/m) temperature (°C) Fluences (>1 MeV) Fluences (>0.1 MeV (neutrons/cm ²) (neutrons/cm ²)		per acou	
				25 × 76 × 76 mm spe	cimens		
A72W26	-25	38	285	2.01E+19	1.24E+20	0.0470	
A72W30	-25	43	283	2.04E+19	1.26E+20	0.0478	
A72927	29	61	283	2.06E+19	1.28E+20	0.0483	
A72W21	30	53	281	2.34E+19	1.41E+20	0.0538	
A72W32	32	66	285	2.31E+19	1.39E+20	0.0530	D
A72W22	60	62	284	2.03E+19	1.26E+20	0.0475	A
A72123	60	74	282	2.32E+19	1.39E+20	0.0532	D
A72W29	60	70	286	2.33E+19	1.41E+20	0.0536	
				25 × 152 × 152 mm sp	ecimens		
a 72013	60	81	287	1.68E+19	1.11E+20	0.0415	
A721210	75	101	289	1.64E+19	1.12E+20	G.0410	
A72011	75	133	286	1.52E+19	1.01E+20	0.0376	
A72015	75	114	289	1.54E+19	1.03E+20	0.0379	
A72U16	76	102	289	1.53E+19	1.02E+20	0.0377	
A72006	90	160	289	1.65E+19	i.13E+20	0.6413	C.D
A72W18	90	132	287	1.42E+19	9.21E+19	0.0345	
				33 × 152 × 152 mm sp	ecimens		
A72009	90	120	289	1.67E+19	1.14E+20	0.0418	

Table 9. Irradiated crack-arrest tonghness data for the weld-embrittled type specimens from weldment 72W (normalized $RT_{MDY} = 57^{\circ}C$). The average fluence and irradiation

temperatures were 1.88 × 10¹⁹ neutrons/cm² (>1 MeV) and 286°C, respectively

^aOne or more letters for a specimen indicate that the test results did not meet one of the minimum lengths of the ASTM E 1221-88 validity criteria. The letters correspond to those in Table 2 of ASTM E 1221-88, which can be paraphrased as:

1.85E+19

1.85E+19

A,B - unbroken ligament too short.

C - specimen too thin.

100

100

D,E - insufficient crack-jump length.

144

118

NOTE:

A72W14

A72W17

K. - value of stress intensity factor shortly after arrest.

289

289

RTwor - reference nil-ducrility-transition temperature.

0.0462

0.0463

1.26E+20

1:26E+20

Table 10.	Irradiated	crack-arrest	toughness da	ta for	the well	d-embrittle	ed type specimen
from w	veldment 73W	(normalized H	$T_{MDT} = 59^{\circ}C$).	The	average	fluence and	d irradiation
ter	mperatures w	ere 1.93 × 10	19 neutrons/ci	n ² (>1	MeV) and	1 285°C. To	spectively

Specimen	Test	Ke (MPas (m)	Irradiation	Exposure	values	Displacements	Validity ^a
	(*C) (*C)	C) (°C)		(*C) Fluences (>1 MeV) Fluences (>0.1 MeV) (neutrons/cm ²) (neutrons/cm ²)		per atom	
				25 × 76 × 76 mm spec	imens		
A73W37b	-25	29	280	2.38E+19	1.44E+20	0.0548	A
A73W39	-25	41	282	2.13E+19	1.31E+20	0.0497	
A73W31	29	57	283	2.10E+19	1 29E+20	0.0490	p
A73W34	29	60	283	2.08E+19	1.28E+20	0.0484	D
A73W35	29	67	283	2.37E+19	1.42E+20	0.0544	D
A73W33	60	74	282	2.39E+19	1.44E+20	0.0550	D
A73W40	60	81	280	2.36E+19	1.42E+20	0.0542	D
A73W41	60	79	284	2.11E+19	1.30E+20	0.0492	D
				25 × 152 × 152 mm spe	ecimens		
A73W15	60	81	289	1.68E+19	1.15E+20	0.0421	
A73W13	75	150	289	1.69E+19	1.16E+20	0.0424	C,D
A73W18	75	131	287	1.56E+19	1.04E+20	0.0385	
A73W21	75	107	289	1.73E+19	1.14E+20	0.0426	
A73W26	75	88	288	1.58E+19	1.05E+20	0.0387	
A73W23	90	180	288	1.58E+19	1.05E+20	0.0389	B,C
A73W45	90	114	287	1.46E+19	9.45E+19	0.0354	
				33 × 152 × 152 mm spe	cimens		
A73W14	90	159	289	1.71E+19	1.17E+20	0.0429	B,C
A73W17C	100		289	1.90E+19	1.29E+20	0.0474	
A73W51	100	184	289	1.90E+19	1.30E+20	0.0475	B,C

⁸One or more letters for a specimen indicate that the test results did not meet one of the minimum lengths of the ASIM E 1221-88 validity criteria. The letters correspond to those in Table 2 of ASIM E 1221-88, which can be paraphrased as:

A,B - unbroken ligament too short.

C - specimen too thin.

D,E - insufficient crack-jump length.

^bValue of K_s is unrealistically low because remaining ligament is too small.

 $^{\rm C}$ Specimen exhibited tearing behavior when tested at this temperature, then broke without arresting when retested at 30° C.

NOTE

K_a - value of stress intensity factor shortly after arrest.

RT_{MDT} = reference nil-ductility-transition temperature.

In Tables 7 through 10, the results of the crack-arrest tests that were "invalid" relative to the requirements of ASTM E 1221-88 are indicated by one or more letters, a blank implying a "valid" result. To judge the degree of "invalidity," see the detailed results given in Appendix A.

The crack-arrest toughnesses, K_a , as a function of test temperature for the different materials, specimen sizes, and specimen types are plotted in Figs. 15 through 18. Comparisons of the unirradiated and irradiated crack-arrest toughnesses for each of the 72W and 73W welds are shown in Figs. 19 and 20, respectively. These figures also show the unirradiated and irradiated ASME K₁₈ curves for each weldment indexed to their respective RT_{NDT} values. In the plots showing the results of the irradiated crack-arrest testing for weldment 73W, the data point at -25°C below the ASME K₁₈ curve is that of specimen A73W37, described above.

Lower-bound curves to the test results have been plotted as dotted curves in Figs. 19 through 21. The dotted curves are ASME curves that have been shifted downward in temperature until the first data point is encountered. The amount of the shift is shown for both the unirradiated and irradiated conditions. In the case of 72W, the temperature shift downward from the normalized curve for the irradiated specimens is 8 K smaller than that for unirradiated specimens. For the 73W specimens, however, the opposite is obtained; the downward shift for the irradiated specimens results is 11 K greater than that for the unirradiated specimens. With all the uncertainties involved, the differences between the downward shifts are not deemed to be significant. Thus, the preliminary observation is that the shift in K_a due to irradiation is about the same as the shift in the CVN-impact energy at the 41-J energy level, as can be judged by the dotted curves in Figs. 19 and 20.

All the crack-arrest toughnesses, both unirradiated and irradiated, for both welds have been plotted as a function of T-RT_{NDT} in Fig. 21. The irradiated specimen A73W17 described above (whose remaining ligament was too small to yield an accurate result) is not included in this figure. Figure 21 includes a total of 77 unirradiated and 34 irradiated data points, many of which overlap. This figure indicates that the results form a reasonable trend when indexed to RT_{NDT} . The normalized RT_{NDT} s (Table 6) have been used to index the irradiated data. When shifted by the normalized CVN ΔTT_{41-3} , the ASME curve is a conservative estimate of the irradiated crack-arrest toughness of the 72W and 73W weldments in the transition region to approximately 40 K above RT_{NDT} . At temperatures below RT_{NDT} , there seems to be a smaller K_{Ia} margin between the lower-bound curves and the ASME K_{Ia} curves.

The shape of the lower-bound curves shown dotted in Figs. 19 and 20 for the data obtained in Phase 1 of the Sixth Irradiation Series do not seem to have been altered by irradiation. The CVN-impact energy curve for both welds, especially for 73W, changed shape when irradiated to a







Fig. 16. Detailed crack-arrest toughness K_a vs test temperature for the unirradiated 73W weld showing the different specimen sizes and types used.



Fig. 17. Detailed crack-arrest toughness $K_{\rm a}$ vs test temperature for the *irradiated* 72W weld showing the different specimen sizes and types used.



Fig. 18. Detailed crack-arrest toughness K_a vs test temperature for the *irradiated* 73W weld showing the different specimen sizes and types used.







Fig. 20. Unirradiated and irradiated crack-arrest toughness $K_{\rm a}$ vs test temperature for the 73W weld.

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MARY C





fluence level of 1.51×10^{19} neutrons/cm² (>1 MeV), (see Ref. [3] and Fig. 14) Many empirical correlations exist between the CVN-impact energy and fracture toughness of unirradiated ferritic steels (Ref. [9]). If such a relationship also exists for irradiated ferritic steels and if the irradiated CVN curve changes shape, then the irradiated fracture toughness curve may also change shape correspondingly. One of the objectives of the Fifth and Sixth Irradiation Series research programs is to investigate whether irradiation can induce such a shape change in the toughness curves. Indeed, such a change has been observed in the lowerbound curve to the irradiated initiation toughness K_{Je} data for the 73W weldment. Although no such change in shape has been observed in the K_a data obtained in tests to approximately 40 K above RT_{NDT}, further tests at higher temperatures are needed to ascertain whether a similar change in shape occurs in the arrest toughness curve. Moreover, statistical analyses of all the results will be performed to examine that question.

Successful cleavage crack initiation occurred in 35 of the 36 specimens tested. Such a success rate is unusual even for testing unirradiated specimens. Moreover, successful unstable crack initiation occurred in weld-embrittled-type specimens at test temperatures 40 K above NDT. A test temperature of approximately 20 K above NDT is generally considered to be the upper limit for a successful unstable crack to initiate in unirradiated weld-embrittled-type crack-arrest specimens for the steels and specimen thicknesses used here. It is likely that the radiation-induced increases in strength of the test material and the brittle-weld crack-starter material allows for a higher crack-driving force to enable testing at higher temperatures.

4. COMPARISON WITH OTHER DATA

Reference [2], which describes an Electric Power Research Institute (EPRI) project, contains almost all the published data on the effect of irradiation on crack-arrest toughness. Four steels were tested in the EPRI project: two welds and two plate materials. Both the plates and the welds included a low-copper as well as a high-copper steel. The total number of irradiated data points from all materials is 34. The results of the irradiated crack-arrest toughness tests were compared with the unirradiated data by shifting the irradiated data downward in temperature to achieve an approximate data coincidence. The EPRI downward shift is approximately comparable to the lower-bound shift that we used (obtained by shifting the ASME K_I curve downwards in temperature until the first data point was encountered). The EPRI shift in K_{Ia} due to irradiation for high-copper materials is less than $\Delta TT_{A1-J},$ whereas the shift in K_{IA} for the low-copper materials is slightly greater or equal to ΔTT_{A1-J} (in the EPRI program, the CVN shifts were also evaluated in several other ways). A similar trend is obtained in this study, although the range of copper contents is quite different. In the EPRI program, the copper contents ranged from approximately 0.03 to 0.23 % by weight. In our study, the copper contents for the two materials were 0.23 and 0.31% and the normalized ATT_{A1-J} values were 80 and 93 K, the shifts of the lower-bound Ka curves were 88 and 82 K for the 72W and 73W welds respectively. We also increased the maximum values of irradiated crack-arrest toughness obtained when compared with those of the EPRI program from approximately 130 to 185 MPa √m. Irradiated crack-arrest toughness data for both welds in our study were also obtained at higher temperatures with respect to the normalized adjusted RTNDT (up to 40 K above RTNDT) than were the EPRI data for the high-copper weld (up to approximately the adjusted RTNDT).

5. SUMMARY

Crack-arrest testing of high-copper, submerged-arc welds was performed on unirradiated and irradiated weld-embrittled-type specimens 25- and 33-mm thick. Most of the crack-arrest test results are either valid or only marginally invalid according to ASTM E 1221-88. The 35 data points obtained by testing the irradiated crack-arrest specimens have approximately doubled the known data base of irradiated crack-arrest toughness and extended the data base coverage to higher levels of crack-arrest toughness and temperature relative to RT_{NDT} . Preliminary observations are:

- 1. Values of irradiated crack-arrest toughness K_{Ia} were obtained at temperatures 40 K above the irradiated RT_{NDT} of the welds. This accomplishment is experimentally significant because a temperature of 20 K above RT_{NDT} is generally considered to be the limit for obtaining useful results with the unirradiated weld-embrittled type of crack-arrest specimen.
- 2. The shifts of the lower-bound K_a curves for the 72W and 73W welds are approximately the same as the corresponding 41-J CVN-implict energy level shifts.
- 3. The ASME K_{Ia} curve, when shifted by ΔTT_{41-J} , is a conservative estimate of the irradiated crack-arrest toughness for welds 72W and 73W in the transition region 40 K above RT_{NDT} . At temperatures below RT_{NDT} , a smaller margin of toughness is apparent between the lower-bound curves and the ASME K_{Ia} curves.
- The shape of the lower-bound curves compared to those of the ASME K_{1a} curves were apparently unaltered by irradiation for the temperature range covered by the tests.

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APPENDIX A

DETAILED CRACK-ARREST SPECIMEN DATA AND TEST RESULTS

Flow and Processing of Crack Arrest Data

The appendices document for archival and quality assurance purposes various aspects of the crack-arrest data. Appendix A traces the flow and processing of data on the crack-arrest specimens and also gives detailed crack-arrest specimen dimensions and results. The BASIC computer code "CA_TEST" used to process the test data is listed in Appendix B, and the Young's moduli used are discussed in Appendix C. Typical output from CA_TEST for weld-embrittled and duplex-type crack-arrest specimens is shown in Appendix D. The load vs crack mouth opening displacement (CMOD) charts obtained during the test and a photograph of the fracture surface for each irradiated specimen is reproduced in Appendix E.

It is not possible to legibly present the voluminous data about each specimen in one table without resorting to foldout pages. Thus, to limit table widths to a single page, the data are logically grouped and the specimen identification is used to the tables together. To facilitate their use, the crack-arrest data tables given in the appendices are sorted by specimen identification (in the main body of the text they are sorted by test temperature).

The dimensions of each crack-ar.est specimen are measured and recorded on data sheets, a sample of which is shown in Fig. A.1. The data on crack-arrest specimens are then recorded on a Hewlett Packard Series 200/300 computer. The BASIC computer code used for this purpose is CA_TEST, (grack arrest test), which was specifically written to process the crack-arrest test data. A listing of CA_TEST is included in Appendix B. The pretest input data consist of the measured specimendimensions, test temperature, yield strength, and Young's modulus at the test temperature. The measured specimen dimensions, for the unirradiated and irradiated crack arrest tests are given in Tables A-1 and A-2, respectively. The yield strength and Young's moduli used have not been included in these tables. The yield strength for each of the 72W and 73W weldments are presented as Figs. 10 and 11 in the main body of the test, and Young's moduli are given in Appendix C.

The ASTM E 1221-88 test procedure prescribes for weld-embrittledtype specimens only a series of loading and unloading cycles. The yield strength and Young's modulus are used to calculate the load increment for the loading and unloading cycles for weld-embrittled-type specimens as well as the validity criteria. Young's modulus is also used to evaluate the stress intensity factors.

The pretest output for CA_TEST is an echo of the input data. CA_TEST also calculates and prints the load increments for the loading and unloading steps in terms of clip-gage CMOD. For duplex specimens, no loading and unloading cycles are prescribed in ASTM E 1221-88, but a single cycle with a CMOD approximately equal to that for weld-embrittled specimens is performed to seat the clip gage and generally shake down the test equipment. After the rapid crack propagation and arrest event, the specimen is heat-tinted, chilled, and broken open. The length of the remaining ligament is measured at three locations by the procedure given in paragraph 8.6.2 of E 1221-88. These three lengths as well as the clipgage CMODs at the four points on the load vs CMOD chart as prescribed in ASTM E 1221-88 are provided as posttest input to CA_TEST.

The maximum and minimum loads registered just before and after the rapid crack propagation event are also recorded. During the development of ASTM E 1221-88, the ratio of these loads was one of the validity requirements. Although the latest revision of ASTM E 1221-88 does not prescribe this ratio, it is still recorded for consistency with the large number of existing files. Experience has also shown that the relative load drop is an approximate indicator of the crack-jump length at the time of the test. A small drop in load is an indication that the crack has not propagated a significant amount. At the other extreme, if the load drops to almost zero, it is likely that the remaining ligament is small. All the posttest data are given in Tables A-3 and A-4 for the unirradiated and irradiated specimens respectively.

The output of CA TEST echoes the posttest data input and calculates K_e and K_o as well as all the validity criteric prescribed in ASTM E 1221-88. The output also indicates which of the criteria have been met or not met. Typical output from CA TEST for both a weld-embrittled and a duplex specimen is given in Appendix D.

The detailed values used to determine the validity of the crackarrest results are given in Tables A-5 through A-8. Tables A-5 and A-6 are for the unirradiated 72W and 73W weldments respectively, and the corresponding information for the irradiated weldments is in Tables A-7 and A-8. In these four tables, the first row for each specimen contains the measured values required to judge the validity of the results. Below these values are the corresponding minimum values as prescribed in ASTM E 1221-88. Validity criteria are evaluated for weld-embrittled specimens (4 criteria) and duplex-type specimens (5 criteria).

In the right-hand column, the presence of one or more letters indicates that the specimen did not meet the validity criteria of ASTM E 1221-88. The arrested crack front is rarely straight or square. Thus, to allow for uncertainties in the measurement of the remaining ligament, the measured values were increased by 10% before they were compared with the ASTM criteria.

Other values included in the tables are the specimen type (whether weld embrittled or duplex) and nominal specimen thickness B and overall width (the height is equal to the width). Other parameters included are test temperature and stress intensity factors just before the rapid crack propagation event K_o (a measure of the crack-driving force) and that shortly after arrest K_a .

During the processing of the data, it was not clear how the information will be eventually presented. To maintain some flexibility, the data are maintained in the relational data base computer code PARADOX. Such a data base allows any of the information stored in it to be selected, sorted, and formatted for presentation. All data (including that from the Hewlett Packard Series 200/300 computers) are maintained in PARADOX on an IBM-compatible computer. The tables in this report have been prepared by using a RDB computer code. The RDB code has also proven useful in the evaluation of the data. When the data are examined from different viewpoints, input data errors are often noticed.

Specimen	В	B _N	2H	Wt	Fr	Rs	Rt	Tu	τv	Я
A72W01	25.40	19.25	152.37	152.45	6.35	10.72	40.18	28.94	9.00	21.54
A72W02	33.05	24,73	152.37	152.45	6.35	10.69	40.24	28.89	0.00	12 08
A72W03	25.40	17.86	152.40	152.45	6.35	10.77	40.26	29.43	0.00	11 61
A72₩04	33.05	24,94	152.37	152.45	6.35	10.73	40.26	28.90	0.00	11 51
A72W05	25.40	17.86	152.40	152.45	6.35	10.73	40.18	29.39	0 00	11.51
A72W07	33.02	25.12	152.37	152.43	6.35	10.67	40.33	28.91	0.00	13 52
A72¥08	25.43	17.96	152.40	152.43	6.35	10.66	40.19	30.28	0.00	11.04
A72W12	33.05	25.07	152.40	152.43	6.35	10.68	46.32	28.77	0.00	11.45
A72W19	33.02	25.04	152.37	152.43	6.35	10.59	40.26	29.37	0.00	11 54
A72W20	25.37	18.15	152.43	152.43	6.35	16.47	40.52	28.68	0.00	11.60
A72W24	25.35	19.13	76.17	76.20	3.18	3.95	21.61	11.75	0.00	11.49
A72W25	25.35	19.15	76.17	76.17	3.18	3.77	21.54	11 SR	0.00	11.66
A72W28	25.27	19.10	76.17	76.17	3.18	3.95	21.58	11 86	0.00	11 53
A72W31	25.27	19.20	76.17	76.17	3.18	3.95	21.60	11.80	0.00	11 61
A72W34	25.40	19.15	76.15	76.20	3 18	3.89	21.44	12.37	0.00	11.61
A72W35	25.40	19.08	76.20	76.23	3.18	3.93	21.66	11.86	0.00	11.57
A721436	25.40	19.15	76.17	76.23	3.18	3.85	21.62	12 12	0.00	11 53
A72W37	25.40	19.02	76.17	76.23	3.18	4.00	21 68	12 03	0.00	11 59
A72W38	25.37	19.05	76.20	76.20	3.18	4.01	21.48	12.22	0.00	11 50
A72W39	25.40	19.00	152 37	152.40	6.35	10.62	40 44	28 56	0.00	11 52
A72W40	32.99	24.74	152.37	152.40	6.35	10 73	40.56	29.66	0.00	11.00
A72W41	25.4	19.00	152.37	152.43	6.35	10.74	40.50	28.39	0.00	11.55

Table A-1. Measured dimensions in millimeters of the unirradiated crack-arrest specimen, from the 72W and 73W weldments^a

Specimen	в	By	28	W _t	Pτ	<u>R</u> .s	Rt	Tu	Tν	N
A72W67	25.43	19.00	152.40	152.37	6.35	10.62	40.44	28.42	0.00	11.50
A72144	25.40	19.02	152.37	152.37	6.35	10.63	40.38	28.52	0.00	11.47
A721245	32.99	24.74	152.40	152.40	6.35	10,56	40.40	28.89	0.00	11.41
A72146	25.43	18.92	152.37	152.37	6.35	10.57	40.50	28.43	0.00	11.49
A72W47	33.02	24.74	152.37	152.40	6.35	10.70	40.51	28.52	0.00	11,50
A72W48	25.40	18.87	152.37	152.40	6.35	10.69	40.55	28.33	0.09	11.38
A72257	32.97	24.79	152,40	152.40	6.35	9.40	41.13	27.07	42,67	1.43
A72062	32.99	24.82	152.40	152.40	6.35	9.53	41.34	26.76	42.67	1.15
A72W63	32.99	24.84	152.35	152.37	6.35	9.55	41.39	26.50	42.67	1.04
A72W64	33.02	24.79	152.45	152.35	6.35	9.48	41.16	26.70	42.67	1.08
A72W65	33.05	24.79	152.40	152.40	6.35	9.53	41.38	26.89	42.67	1.17
A72W66	33.02	24.77	152.40	152.43	6.35	9.49	48.34	26.55	42.67	1.06
A72¥68	32.99	24.79	152.40	152.40	6.35	9.45	41.33	27.22	42.67	1.10
A72W71	33.02	24.79	152.43	152.43	6.35	9.53	41.25	27.06	42.67	1.06
A72W73	33.05	24.82	152.43	152.40	6.35	9.46	41.20	30.19	42.67	1.13
A72W8 :	50.85	46.99	203.23	197.74	14.29	7.90	48.66	40.18	0.00	11.34
A72W84	50,90	46.99	203.20	197.71	14.29	7.76	48.74	40.19	0.00	11.33
A72W85	50.84	38.13	203.33	197.76	14.27	7.87	50.30	54.35	0.00	11.86
A73W01	25.45	19.02	76.15	76.15	3.18	3.65	21.65	11.82	0.00	11.52
A73W02	25.45	19.02	76.17	76.12	3.18	3.57	21.84	11.52	0.00	11.60
A73W03	25.45	19.10	76.15	76.12	3.18	3.70	21.58	11.74	0.00	11.52
A73204	25.45	19.00	76.17	76.12	3.18	3.76	21.26	12.18	0.00	11.52
A73W05	25.45	19.13	76.17	76.10	3.18	3.84	21.64	11.70	0.00	11.51
A73906	25.45	19.02	76.17	76.02	3.18	3.79	21.64	11.69	0.00	11.56
A73W07	32.97	24.93	152.35	152.27	6.35	9.54	41.12	26.74	43.18	2.37
A73W08	32.97	24.97	152.37	152.37	6.35	9.60	41.15	26.83	43.18	2.45
A73W09	32.94	24.94	152.35	152.37	6.35	9.57	41.11	27.00	43.18	2.36
*72110	32.00	26 88	152 37	152.40	6.35	10.41	40.29	34.39	0.00	11.52

Table A-1 (continued)

Specimen	В	By	28	w _t	₽r	Rs	Rt	Τu	Ŧv	N
A73W11	25.43	18.97	152.43	152.43	6.35	10.58	40.14	34.65	0.00	11.59
A73W16	25.40	:8.95	152.35	152.37	6.35	10.50	40.30	34.59	0.00	11.56
A73W20	25.40	18.92	152.35	152.40	6.35	10.41	40.29	34.03	0.00	11.60
A73W22	33.05	24.78	152.40	152.37	6.35	10.37	40.14	34.60	0.00	11.52
A73W24	32.99	24.80	152.37	152.37	6.35	10.38	40.39	34.62	0.00	11.45
A73W25	25.40	19.02	152.35	152.40	6.35	10.52	40.45	34.19	0.00	11.57
A73W27	32.99	24.77	152.37	152.40	6.35	10.45	40.19	34.28	0.00	11.53
A73W28	25.40	18.97	152.35	152.40	6.35	10.43	40.43	34.38	0.00	11.58
A73W29	33.02	24.77	152.37	152.40	6.35	10.30	40.33	34.31	0.00	11.51
A73¥30	25.40	18.95	152.43	152.45	6.35	10.51	40.30	34.48	0.00	11.59
A73W32	25.40	19,10	76.17	76.23	3.18	3.83	21.60	12.13	0.00	11.53
A73W36	25.40	19.02	76.17	76.23	3.18	4.02	21.56	11.81	0.00	11.49
A73W38	25.40	19.15	76.17	76.23	3.18	3.83	21.53	12.11	0.00	11.49
A73W42	25.40	19.08	76.20	76.23	3.18	3.85	21.46	12.17	0.00	11.55
A73W43	25.40	19.02	152.37	152.40	6.35	10.61	40.39	28.61	0.00	11.51
A73W44	33.05	24.79	152.40	152.40	6.35	10.61	40.46	29.62	0.00	11.52
A73W46	33.02	24.82	152.37	152.40	6.35	10.62	40.33	28.67	0.00	11.70
A73W47	25.43	18.95	152.40	152.40	6.35	10.73	40.41	28.65	0.00	11.48
A73W48	25.40	18.87	152.37	152.37	6.35	10.59	40.46	28.76	0.00	11.57
A73W49	33.02	24.74	152.32	152.37	6.35	10.59	40.43	28.52	0.00	11.37
A73W50	25.43	19.08	152.37	152,40	6.35	10.71	40.37	28.74	0.00	11.57
A73W52	25.43	19.05	152.40	152.37	6.35	10.76	40.59	28.64	0.00	11.55
A73W75	49.53	37.26	203.23	203.28	0.84	8.43	55.16	37.10	66.06	1.50
A73W85	33.05	24.87	152.32	152.43	6.27	9.58	41.30	27.97	42.57	1.34
A73W86	33.02	24.89	152.43	152.45	6.27	9.51	41.27	27.84	42.77	1.39
A73W87	32.99	24.89	152.45	152.43	6.27	9.53	41.26	28.69	44.19	1.38
A73W88	32.97	24.99	152.50	152.43	6.27	9.46	41.25	28.72	42.42	1.37

Table A.1 (continued)

^aSee Fig. A.1 for definitions of nomenclature used in this table.

Specimen	В	B _N	2H	Wt	Pr	Rs	Rt	Tu	N
A 72W06	25.40	17.88	152.40	152.40	6.35	10.51	40.22	29.56	11.65
A72000	33 02	24.82	152.40	152.40	6.35	10.50	40.23	30.13	11.51
A72010	25 40	17.93	152.40	152.40	6.35	10.52	40.31	28.85	11.53
x72011	25 40	17.83	152.40	152.40	6.35	10.43	40.29	29.35	11.51
A79013	25 40	17.83	152.40	152.40	6.35	11.87	41.68	28.79	11.39
A 2 21/2 W 1 J	33.02	25 02	152.40	152.40	6.35	10.49	40.27	28.92	11.51
A72015	25.40	17.81	152.40	152.40	6.35	10.54	40.34	28.96	11.55
A72016	25.40	17.83	152.40	152.40	6.35	10.48	40.29	29.37	11.56
A72W10	33.02	24 79	152.37	152.40	6.35	10.49	40.34	28.97	11.15
A701110	25.40	18 01	152.40	152.40	6.35	10.55	40.34	28.89	11.51
A72W10	25.40	19 08	76.20	76.20	3.18	3.59	21.45	11.60	11.59
A70000	25.40	19 10	76.20	76.20	3.18	3.73	21.36	11.91	11.48
RIZWZZ	25.40	19 10	76.20	76.20	3.18	3.82	21.48	11.81	11.48
R/LWL)	25.40	10 10	76.20	76.20	3.18	3.94	21.62	11.76	11.56
AIZWZO	25.40	10 13	76 20	76.20	3.18	3.91	21.54	11.80	11.51
AIZWZI	23.40	10.08	76.20	76 20	3.18	3.64	21.35	11.81	11.55
A/2W29	25.40	10.15	76 20	76.20	3.18	3.98	21.43	11.90	11.66
A72W30 A72W32	25.40	19.13	76.20	76.20	3.18	3.60	21.43	11.78	11.54

Table A-2. Measured dimensions in millimeters of the *irradiated* weld-embrittled crack arrest specimens from the 72W and 73W weldments^a

Tsole A-2 (continued)

pecimen									
2 1116 1	07 50	18.92	152.40	152.40	6.35	10.35	40.33	34.58	IT.>>
71461	33 02	74 77	152.40	:52.40	6.35	10.45	40.32	34.54	11.52
DIME /	30.00	18 05	152.60	152.40	6.35	10.41	40.35	34.59	11.62
CIMS/	04.02	22 76	152 40	152.40	6.35	10.41	40.35	34.49	11.53
1.3WE	20.00	10 07	152 40	152 40	6.35	10.27	40.41	34.48	31.60
73418	00.02	06 81	152.40	152.40	6.35	10.13	40.33	34.49	11.52
E GERE E	35.60	18 90	152.40	152.40	6.35	10.31	40.32	34.52	11.64
76061	25 40	18.87	152.46	152.40	6.35	10.38	40.21	34.67	11.55
12mcs	25 40	19.08	76.20	76.20	3.18	3.80	21.63	12.12	31.55
E E E E E E E	07 50	19.08	76.20	76.20	3.18	3.75	21.62	12.10	11.49
CCMC I	55 AD	50 61	76.20	76.20	3.18	3.78	21.65	11.99	11.48
toCMC /	04.10	10 10	76.20	76.20	3.18	3.87	21.59	12.18	11.58
13W32	0tb . C7	10.05	76.20	76.20	3.18	3.83	21.57	12.35	11.59
73437	0.0.02	10.13	76.20	76.20	3.18	3.80	21.65	12.07	11.53
73439	22.40	10.00	76. 20	76.20	3.18	3.75	21.60	12.13	11.54
13440	0.0.0	10.00	00 94	76 20	3.18	3.79	21.60	12.05	11.56
73W41	22.40	19.02	152 60	152.40	6.35	10.68	40.45	28.74	11.51
CDW2/	33.02	24.71	152.40	152.40	6.35	10.50	40.55	28.57	11.47

Specimen	W1	W2	83	F1	P.2	P3	P4	Peusk	Pare
A72¥01	33.01	30.94	33.66	0.056	0.163	1.351	1.369	52	4
A72W02	28.64	27.53	29.22	0.038	0.147	1.369	1.443	71	
872903	42.67	62.52	42.70	0.051	0.165	1.405	1.455	54	17
872904	37.80	25.65	31.22	0.025	0.122	1.367	1.445	6.8	
A72W05	36.73	36.13	42.57	0.046	0.165	1.265	1.326	54	- 11
A72W07	34.68	34.73	34.43	0.061	0.163	1.367	1.433	73	
A72W08	35.86	34.64	34.81	0.030	0.145	1,283	1.346	5.2	1. 6
A72W12	33.03	34.23	33.78	0.038	0.312	1.600	1.651	77	1.1
A72W19	33.09	36,33	3.3.54	0.051	0.142	1.224	1.326	71	
A72W20	39.10	35.34	42.45	0.000	0.000	0.714	0.767	29	
A72924	10.58	10.37	10.35	0.020	0.058	0.742	0.815	27	
A72925	12.50	11.65	14.60	0.046	0.109	0.889	0.932	2.6	- 2
A72W28	9.82	10.54	9.45	0.018	0.079	0.762	0.828	33	
A72V31	13.26	13.46	15.78	0.020	0.208	1.168	1.219	51	1
A72¥34	14.77	13.69	15,95	0.013	0.043	0.732	0.737	2.8	2
A72¥35	12.02	12.56	12.14	0.008	0.020	0.554	0.630	25	6
A72W36	13.67	14.84	11.82	0.010	0 069	0.737	0.800	31	
A72W37	14.20	14.47	14.15	0.008	0.015	0.610	0.709	2.6	
A72W38	11.65	11.74	11.15	0.020	0.028	0.620	0.699	2.8	
A72W39	13.8?	14.46	13.79	0.030	0.030	0.879	1.095	34	
A72W40	24.79	27.18	23.77	0.051	0.320	1.892	1.956	8.7	
672W41	34.88	34.19	33.12	0.056	0.056	0.927	1.016	30	
A72W43	33.35	31.12	34.07	0.025	0.066	1.184	1.245	4.9	
A72844	34.18	34.21	33.34	0.018	0.043	0.925	1.019	44	2
A72W45	38.29	38.93	6.84	0.058	0.058	0.889	0.980	3.8	2
A72W46	46.44	48.73	42.19	0.056	0.056	0.958	0.980	3.6	8
∆72₩47	33.36	38.38	29.52	0.061	0.097	1.189	1.255	5.6	2
A72W48	34.57	35.19	33.04	0.038	0.076	1.214	1.306	53	
A72W57	54.48	54.34	56.69	0.018	0.000	1.283	1.316	7.8	4
A72W62	37.13	37.39	36.83	0.000	0.000	1.577	1.659	107	8
A72W63	39.81	40.04	41.37	0.005	0.000	1.361	1.427	143	1
A72W64	37.49	35.73	29.64	0.036	0 000	1.275	1.306	75	10
A72W65	34.33	34.35	38.15	0.010	0.000	1.509	1.554	84	8

Table A-3. Posttest values measured for unirradiated crack-arrest specimens from the $72\rm W$ and $73\rm W$ weldments

Table A-3. (continued)

Specimen	¥1	W2	6.9	P1	P 2	P3	p4	$\mathbf{p}_{\mathrm{sand}}$	Pate
A.72W66	31.37	32.20	30.64	0.018	0.000	1.346	1.402	87	3
A72W68	35.12	35.53	38.25	0.020	0.000	1.359	1.443	8.9	
672971	23.22	22.16	25.71	0.025	0.000	1.392	1.504	106	
A72¥73	48.18	50.45	43.83	0.046	0.000	1.415	1.448	64	
A72W83	59.33	60.56	56.02	0.079	0.079	1.209	1.224	6.6	
A72¥84	53.72	55.41	60.47	0.086	0.178	1.570	1.651	121	4
A72W85	32.98	39.89	28.27	0.102	0.163	1.745	1.816	7.6	6
A73W01	10.14	10.87	9.47	0.051	0.038	0.650	0.699	18	0
A73W02	14.24	12 53	15.76	0.025	0.038	0.732	0.808	27	0
A73W03	7.76	7.89	8.08	0.030	0 397	0.919	0.998	2.6	.0
A73W04	10.36	9.00	11.43	0.030	0.102	0.838	0.909	3.8	
A73W05	8.04	9.48	7.29	0.038	0.193	1.021	1,125	3.8	0
A73W06	12.99	13.00	12.67	0.043	0.254	0.996	1.052	37	9
A73W07	30.21	34.74	31.51	0.025	0.000	1.636	1.783	119	0
A73¥08	39.08	38.93	40.81	0.025	6.000	1.313	1.417	8.8	1
A73W09	37.24	\$9.62	37.05	0.025	0.000	1.283	1.372	9.7	G
A73W10	16.62	16.62	16.31	0.030	0.229	1.605	1.671	6.6	
A73W11	34.95	33.74	36.70	0.051	0.079	1.085	1.107	31	12
A73W16	32,55	34,35	31.80	0.013	0.051	1.143	1.186	48	
A73W20	40.31	40.56	39.90	0.051	0.262	1.542	1.565	49	5.5
A73W22	27.64	26.80	29.91	0.013	0.173	1.651	1.671	54	0
A73W24	14.07	14.66	11,93	0.038	0.361	2.108	2.200	6.8	
A73W25	35.53	35.93	39.76	0.038	0,222	1.781	1.819	48	12
A73W27	34.23	38.29	34.82	0.025	0.125	1.407	1.433	61	0
A73W28	29.54	30.73	28.66	0.051	0.051	0.968	1.003	30	2
A73W29	31.24	26.85	34.02	0.064	0.099	1.402	1.433	6.6	
A73W30	25.60	28.52	23.98	0.025	0.053	1.067	1.118	31	.0
A73W32	6.60	6.91	6.93	0.013	0.058	0.841	0.927	3.7	0
A73W26	13.84	13.75	12.90	0.013	0.043	0.716	0.792	31	. I.,
A73W38	6.98	6.83	6.81	0.020	0.048	0.699	0.876	23	0
A?3W42	8.80	8.09	9.36	0.025	0.053	0.726	0.874	34	0
A73W43	35.72	36.16	36.54	0.056	0.056	0.892	0.968	43	
A73W44	41.77	44.87	42.95	0.064	0.150	1.372	1.433	75	18
A73W46	41.28	39.15	42.24	0.030	0.137	1.433	1.473	73	
A73847	19.58	41.03	39.26	0.056	0.056	0.968	1.031	38	8
A73948	28.27	26.48	29.53	0.051	0.091	1.090	1.166	40	3
A73W49	35.39	33.42	38.65	0.030	0.206	1.516	1.539	64	0

Specimen	¥1	W2	W3	P1	87	F3	P-4	Pmax	Luin
A79850	18.05	18.28	22.52	0.036	0.152	1.435	1.560	6.7	0
A72952	18.32	18.84	17.84	0.051	0.155	1.448	1.567	6.6	
A73575	40.84	37.67	45.12	0.000	0.000	1.613	1.765	181	
473W85	50.63	50.28	49.94	0.000	0.000	1.280	1.313	8.2	
A73W86	34.19	35.10	32.86	0.000	0.000	1.245	1.285	83	1.1
A73W87	30.58	30.94	30.74	0.000	0.000	1.491	1.554	91	1.2
A73W88	42.81	46.58	42.56	0.000	0.000	1.387	1.407	8.2	3

Table A-3. (continued)

NOTES: W1,W2, and W3 = lengths in millimeters of the remaining ligament measured according to ASTM E 1221-88

P1 through P4 = displacements in millimeters measured from the load vs CMOD trace. see Fig. A.J.

 $P_{max},\ P_{min}$ = maximum and minimum loads registered just before and just after the rapid crack propagation event.

Specimen	¥1	₩2	43	P1	P2	P3	1%	Paur	\mathbf{P}_{BAB}
A72W06	66.1	67.45	68.33	0	0 127	1 213	1.265	- 21	16
412W09	46.34	45.59	52.86	0.	0.152	1.246	1,308		
A72910	49.31	49.94	48.44		0.051	0.965	1 003		
A72911	60.43	60.25	62.5		0 159	1.124	1.149	1.8	12
A72W13	43.86	45.26	43.38		0.051	0.81	0.879	28	
A72814	59.19	68.55	56.84		0.14	1.232	1.283	58	29
A72W15	46.14	49.14	45.93		0.089	1.13	1.186	18	3
A72W16	54.8	57.81	52.31		0,102	0.919	0.958	- 1.5	
6.72917	50.94	59.45	50.96		0.064	1,095	1.138	24	1.2
A72W18	52.18	66.86	52.02		0.038	1.138	1.138	3.9	1
∧72¥21	20.35	20.48	19			0.404	0.452	6	
A72¥22	6.682	7,145	7.406			0.957	0.97	2	- 1
A72W23	21.77	22.11	22.23			0.538	0.572	1.0	
A72W26	15.57	14.19	15.03		0.013	0.34	0.406		2
A72W27	18.13	19.6	17.3			0.498	0.549	11-	3
A72₩29	20.23	22.09	19,94	0		0.53R	0.574	41	
A72¥30	15.84	15.11	15.25	0			0.414	6	2
A72¥32	24.77	24.78	24.97			0.434	0.472	10	
A73913	67.24	70.09	60.19		0.051	1.151	1.189		14
A73914	29.69	24.59	33.03		0.218	2.355	2.398	58	1
A7315	45.23	67.6	44.04		0.013	0.828	0.859	16	5
A73W17		0	0	0					
A73W18	48.72	54.82	50.23	0	0.083	1.273	1.285	1.5	5
A73W21	52.97	53.16	56.02		0.032	0.972	0,997	1.2	7
A73W23	43.73	50.62	41.69	-0	0,286	2.007	2.045	6.0	9
A/3W26	56.14	55.65	54.27		0.02	0.77	0.813	12	
A73W31	19,97	22.84	21.64			0.408	0.464	8	
A/3W33	26.34	26.68	26.6			0.464	0.495	10	
A73W34	22.15	21.2	22.88	0		0.414	0.48	10	
A73W35	27.82	25.31	30.01			0.411	0.424	9	1.1
A73W37	2.738	3.18	2.556		6:025	0.732	0.732	1	
A73839	13.66	13.86	14.39			0.362	0.464	13	
A73W40	25.99	26.32	26.62	0		5,508	0.546	10	
A73941	31.13	32.23	29.41			0.432	0.472	9	
A73W45	52.69	52,84	60.23	0	0.07	1.034	1.082	.2.6	1
A73W51	47.38	51.28	42.7	-0.013	0.343	2.019	2.057	41	3

Table A-4. Posttest values measured for unirradiated crack arrest specimens from the 72W and 73W weldments

NOTES :

W1.W2, and W3 = lengths in millimeters of the remaining ligament measured according to ASTM E 1221-88.

P1 through P4 - displacements in millimeters measured from the load vs crack mouth opening displacement trace, see Fig. A.1. - maximum and minimum loads registered just before and just after the

Pmax, Pmin rapid crack propagation event

Specimen	Туре	в	We	ť	£.4	Ko	6	8		-D	-E	Validity
872901	546	25	152	- 14		199	13	33		51		
							3.9	23	3.8	23	28	
ATENOZ	NE.		1.52			20.5	29	29				
A72903	VE	25	152	34	178	234	43	43	25	40	611	
							19	40	3.2	23	28	
A72W04	WE.	33	152	-15	94	209	2.6	28	33	>5	55	
+ 1900V	118		14.7	1.11		101	19	21	17	23	20	
N. C. C. NO. V.	* D		1.26			1.9.4	19	27	22	23	2.5	
A72907	WE.	33	152	-15	103	201	35	3.5	33	49	44	
							1.9	26	21	2.3	- 25	
WASHING .	16.5	2.9	1.92	1.101	109	195	15	35	29	47	47	
A72912	SIE .	33	152		114	215	34	34	33	50	50	
							19	3.2	26	23	29	
A72W19	YE.	13	152		94	180	16	34	13	49	49	
x 79690	UK.	25	155			1.9.6	19	12	1. 11	23		
			1.118			1.2.14	19	10	8	23	8	
A72\$24	VE	25	76	- 30	41	168				3.2	12	
							10	9		23	1.6	
GASMES.	N.E.	-2.9	1.0			1.92		1.1		30	91.	
A72928	WE.	25	7.6			168	10	10	25	13	- 11	
							1.0	-0		23	1.6	
472M31	WE	25	/6		1.08	23/4	1.04	14	. 75	29	- 79	8.3L
A72W34	- WK	2.5	16			168	1.0	28	1.4	23	12	
							10	12	10	23	16	
A72W35	WE -	25	- 76	-60	3	132	12	12	25	31	- 31	
	1.10						10	4		23		
N12W30	W.C.	2.2		ar		199	13	1.1	12	33		
A72937	WC -	25	16	-64	1.6	147	14	14	2.5	2.8	1 38	
							10	9	8	23	11	
A/2W38	WE	25	16	+ 5.9		146	12	12	25	31	- 11	
A72930	VE		15.2	. 445		125	10	in.	14	23	1.44	
							19	14	1.	23	141	
A72W60	KE.	33	152		114	261	2.5	25	13	5.2		8
A TOMAC Y	40					1.000	19	32	- 26	53	42	
N. C. R.H. L	KC.	4.5	125	19.5		148	10	1.2	10	30		
A72W43	WE.	25	152	1.12		190	1 33	13		51		
							1.9	2.0	1.6	23	24	
A/2Waa	ΨE.	52	125	-643	16	151		34		50		
A:2W45	VE	11	152	data		141	1.9	13	-10	23	1.18	
		1.125					19	13	11	25	- 11	
172846	νE	25	1.5.2	-(a)		154	16	16	25	3.8	18	
173463	100		14.0			1.0.2	19	. 19.	15-	23	13	
IN FLORING C	жņ.	1 8ª	1.54		1.1.1.1	100	1.0	20	10	30		
A72W48	WE	25	152		1.000	194	14	34	25	4.9	14	
							1.0	30.00	1.0	3.3	1.1	

Table A-5. Values measured during crack-arrest testing of the unirradiated weldment 72W and 73W

Table A-5. (continued)

Specimen	Type	8	×,	т	Кa	K.o.	٨	8		þ	£.	Sutakita
A72957	DX.	13	152	21	146	216	55	5.6	33	13	13	
							19	50	16.3	- 25 -		
A12862	DX	3.3	152	10	130	267	9.7	37	13	31	- 28	
	1.14	1.1	144		3.25	232	40	60	15	28	28	
NUMBER .	210		1.1.8	1	112		19	39	31	2.5		
A22W66	D.K.	53.	152	1	108	217	3.8	38	33 -	31	- 31	
							1.9	2.9	23	25		
A72965	DX	33	152		125	200	10	30	11	2.5	2.5	
272966	DX	3.5	152	.2	103	229	- 61	31	33	37	37	
CONTRACTOR .							1.9	2.6	21	2.5		
A72W68	DX	3.3	152	5	115	229	36	36	3.3	32	-32	
					10.5	1.5.5	19	2.3	27	25	1.16	
A72W71	DX.	33	102	- 18	3.1	1.07	10	2.0	16	25	16.9	
672673	DX	. 33	152	5	142	730	48	48	33	21	21	0.17
							1.0	50	40	25		
A72W83	WE.	51	203	- 30	85	146	59	20	23	50		
		1.1	1.00		1265	1.72	- 25	17	1.74	20	12	
A72W84	WE.	51	303	9	107	1.2.8	13.7 13.16	24	2.8	22	20	
6.70URS	W.F.	5.1	203	1.18	145	195	34	34	- 51	1.4	50	
		1.122					2.5	22		24	23	
A73W01	WE.	2.5	76	-45	72	151	1.0	10	2.6	33	33	
				1.1.2.2			10		1.1	23	13	
A73¥02	WE.	25	76	-9.5	1/3	172	10	- 124	10	29	1.12	
A 23903	48	55	16	-65	64	203	8		26	35	35	8.8
H.C. S. W.S. S.							10	. 9		23	23	
A73¥04	WE.	25	16	- 3.0	5.7	181	10	10.	26	72	- 32	
							10	11	8	53	- 12	
A73W05	WE	25	7.6	- 10	70	240.9	8	- 8 -	2.0	12	12	(n-n)
473906	U.F.	1.00	16	1.15	83	LRA.	13	13	26	30	30	- N
N. / SHOW			¹⁹ 1		10.0		10	. 6	13	23	19	
A73W07	DX		1.92	5	129	277	32	12	33	3.6	3.6	- 100
							10	42	34	25		
A75¥08	DX	-13	102		119	222	40	40	33	2.8	.78	
i i aumo	11.0		1.1.1.1		114.4	216	1.9	18	13	30		
11.280.4	10A	1. C.	1.28		1.1.8	2.84	19	32	25	2.5		
A73V10	VE.	53	152	-16	12	216	17	- 17	3.3	6.1	61	
							19	15	12	2.3	. 28	
A/3W11	w'E	25	152	- 32	85	159	35	35	25	. 43.	- 33	
. Same	1.1.1.1		100	2.0	2.6	1.2.4	19	1.14	278	4.5	12	
UVJAID	85		134	188	64	1.04	19	19	15	23	12	
A73W20	VE.	25	152	.16	126	203	40	40	25	38	3.8	
							1.9	3.9	3.1	23	- 25	
A/3W22	VE	31	152	16	110	535	28	2.8	13	10	50	
	194	1 days	1.4	1.1	20		12		278	2.3	11	
N/3824	4.5		1.94	-13	-6.9	213	10	20	16	23	54	11.18
A73W25	VE	25	152	.15	141	243	3.7	37	25	41	41	8.6
and a second sec			1.0.00	0 . O .	1.1.1		1.0	4.9	39	23	36	
Table A-5. (continued)

Spenimen	Type	В	$M_{\rm L}$	Ť.	Here.	ĸo	A	6		0	4	Validity
A73W27	VE	33	1.52	- 12	111	202	36	36		42	48	
manne			110			Sec.	1.9		24	23	25	
A. 1 3 W. 2 B	ND		1.25	- 5.1	119	19.0	10	10	122		11	
A73W29	¥E.	3.3	152			205	11	33	13	47	47	
							1.9	24	1.0	23	25	
A73¥30	Y.E	25	192	- 6.6	11	161	26	26	2.5	5.2	52	
673932	VE	2.5	26	1.64	56	192	14	14	2.5	23	36	
							10	1	5	23	20	
A73W36	WE	2.5	76	- 59	22	168	1.6	14	25	29	29	
CONTRACTO				1. 1.11	1.0	1.141	1.0	11	. 9	23	15	
N73836	*1	42	7.9	- 0.8		160	10	1	25	36	36	<u>A</u>
A73842	WE.	25	. 76		58	166			2.5	34	34	
							10	1	6	23	14	
A73843.	WE	25	152	-3.5	10	1.42	36	36	25	42	. 167	
Londy 1	and the second second	44.1	144		1.01.00		19	-12	10	23	11	
N7.2868.9	×1.		1.3.6		163	-200	34	76.3	3.3	3.9		
A73846	WE .	35.	152	-15	124	219	21	41	33	43	43	
							19	3.8	10	23	2.9	
A73847	wE.	2.5	152	-55	85	156	-60	40	- 25	43	43	
A 7 3136 B	erie .	1.1	in state in the				1.9	1.17	13	23	14	
121.3840	**	1.5	1.16	1.14		1.18	10	13	. 17	20	1.20	
A73849	98	- 33	152	-15	117	222	36	36	33	48	48	
							1.9	3.3	27	23		
A73850	¥.E	25	125		8.0	221		20	2.5	66	64	
A TAMAT	L/R	5.6	140			10.64	19	1.5	12	23	- 29	
er canore			1.76	1.64		2.1.2	19	10	- 11	23	28	
A73W75	DX	51	:03		1.07	229	43	101	60	111	- 41	
								1.1	1.9	22		
A73W85	DX	33	152		137	23.4			33	18	18	(C. D.
								1.9	76.7	3.8	. 25	
A73986	DX		1.5.2	- B (101	209	3.64	3/4	3.3	3.4	- 34	
A73¥87	DX	13	162		1111	10.7	· Sec.	1.9	22	20	25	
						24.2	1.15	10	3.2	26	36	
A73¥88	DX	3.3	152	. 15	132	228	lite	Tills	33	2.5	25	
								19	. licia	36	25	

NOTES:

Type:

-- either WE (weld-embrittled) or DX (duplex) + nominal overall srecimen size, in millimeters, see Fig. 2 B. Wi

* test temperature in "C

= stress intensity factor at crack arrest and at crack initiation, respectively. In MPs+/m Ks Kp

The letters A. B. ... E are the letters used in the ASTM E 1221-88 validity criteria. The top row of numbers are the measured values for each criterion, and those below are the minimums specified in the standard. The remaining ligament has to fulfill two criteria, and thus the same value is repeated for A and B. The same is true for the crack jump.

- A.B ... length of remaining ligamont, millimoters.
- * specimen width, miliimeters.

D.E « crack jump, millimeters. For duplex specimens, only D is applicable.

- On or more letters in this column indicates that the specimen did not meet the validity criteria of E 1221-88. To allow for the uncertainty in the measurement of the crack front, the measured values were increased by 10% before comparing them. Validity with the criteria in the standard

Specimen	Туре	В	W _t	т	Ke	Ke	A	8	С	D	E	Validity
A72W06	WE	25	152	90	160	182	67	67	25	15	15	Ċ,D
			160	90	120	176	48	48	33	34	34	
A72W09	WE	33	106				19	2.8	23	23	24	
A72W10	WE	25	152	75	101	155	49	49	25	34	34	
			160	2.5	133	163	61	61	25	22	22	
A72W11	A.F.	25	125				19	35	2.8	23	12	
A72W13	¥E.	25	152	60	81	130	44	1 3	25	38	38	12.5
			165	100	144	177	62	62	33	22	22	
A72W14	W.E.	33	196	100			19	41	33	23	14	
A72W15	VE	25	152	75	114	177	47	47	25	36	36	
		4.6	169	26	102	138	55	55	25	28	28	
A72W16	WE	6.0	100				19	50	16	23	8.5	5
A72W17	WE	33	152	100	118	168	54	54	33	29	29	
			1.00	00	122	185	57	57	25	26	26	
A72V18	WE	25	195	90	136	100	19	34	27	23	16	
A72¥21	VE	25	76	30	53	98	20	20	25	23	23	
11/2812					10	550	9.6	5.2	4.2	23	36	A
A72W22	WE	25	76	60	0.2	229	9.5	7.4	6	23	23	
* 701703	UE	25	26	60	74	129	22	22	2.5	21	21	D
N/EWES	*5	**	10.00				9.5	10	8.3	23	7.	3
A72W26	WE	25	76	- 25	38	81	15	15	22	28	20	4
			76	20	61	121	18	18	25	25	25	
A72W27	WE	4.5	19	**			ç.5	7	5.0	23	6.	1
A72W29	WE	25	76	60	70	129	21	21	25	22	22	
					1.3	0.7	9.6	9.5	25	28	28	2
A72W30	WE	25	76	- 25	4.5		9.5	3.1	2.5	23	3.	5
A72U12	VE	25	76	32	66	105	2.5	25	25	18	1.8	D
NIE BOS							9.6	8.2	6.5	23	4.	6. 0.10
A73W13	WE	25	152	.75	150	168	60	19	41	33	23	12
	LIP	55	152	90	159	325	2.9	29	33	48	48	B,C
A73W14	W.D.	30	1.7.8					19	46	37	23	44
A73W15	W.E.	25	152	60	81	125	46	46	25	32	32	6.3
			160	100				1.9	**			e
A73W17	WE.	33	755	100								
A73W18	WE	2.5	152	75	131	182	51	51	25	26	20	5 1/6
2 1 2 1 2 1 2 2	110	0.5	160	28	107	166	67	19	25	24	2	2. 1.4
NIDWEL	W 2.	20	196		***	1.4.4		19	21	17	2	3 8.4
A73W23	WE.	2.5	152	90	180	263	45	45	25	3.2	3	2 B,C
Contenado de la						114		19	60	48	2	3 29
A73W26	WE	5.2	152	75	66	110	23	19	14	11	2	3 5.4
A73W31	WE	25	76	29	57	98	2.2	22	25	21	2	1 D
								9.5	5.7	4.6	2	3 3.7
A73W33	WE	25	76	60	74	110	21	27	25	16	1	0 D
A73934	UP	25	76	29	60	100	22	22	25	21	2	1 D
and a state of the		2.0						0 5	6 0	5	0	3 3 8

Table A-6. Values measured during crack-arrest testing of irradiated weldment 72W and 73W

Specimen	Type	В	W _t	Т	Ka	Ko	A	В	C	D	E	Validity
A73W35	WE	25	76	29	6.7	99	2.8	2.8	25	15	15	D
A73W37	WE	2.5	7.6	- 2.5	29	171	2.8	9.5	25	6.2	23	3.7 A
A73W39	ΨE	23	76	- 25	41	8.8	14	14	25	29	23	10
A73¥40	WE	25	76	60	81	121	26	26	25	16	23	2.7 D
A73W41	WE	2.5	76	60	7.9	103	31	31	25	12	23	5.8 D
A73¥45	WE	25	152	90	114	158	55	55	25	28	23	4.2
A73W51	WE	33	152	100	184	274	47	47 19	33 63	19 36 50	23 36 23	10 8,C 31

NOTES :

- either WE (weld-embrittled) or DX (duplex) Type:

B, Wg - nominal overall specimen size, in millimeters, see Fig. 2

- test temperature in "C T

- stress intensity factor at clack arrest and at crack initiation. Ka Ko respectively, in MPa. ...

The letters A. B. ... E are the letters used in the ASTM E 1221-88 validity criteria. The top row of numbers are the measured values for each criterion, and those below are the minimums specified in the standard. The remaining ligament has to fulfill two criteria, and thus the same value is repeated for A and B. The same is true for the crack jump.

- A.B = length of remaining ligament, millimeters.
- C . specimen width, millimeters,
- D.E = crack jump, millimeters. For duplex specimens, only D is applicable

Validity = On or more letters in this column indicates that the specimen did not meet the validity criteria of E 1221-8. To allow for the uncertainty in the measurement of the crack front, hte measured values were increased by 10% before comparing them with the criteria in the standard.

sable A.6 (continued)

Crack arrest data sheet



Fig. A-1. Comparison of values of Young's moduli as calculated from an American Society of Mechanical Engineers table and an Electric Power Research Institute expression. APPENDIX B

LISTING OF BASIC COMPUTER CODE CA_TEST

Documentation of Computer Code CA TEST

The listing of the computer code CA_TEST has been included solely for purposes of QA. It details in a concise manner the computational details that were performed with the crack arrest data. Moreover, the methodology in crack-arrest testing is still evolving. The standard test method used in this work, ASTM Test for Determining Plane-Strain Crack-Arrest Fracture Toughness, K_{Ig} , of Ferritic Steels (E 1221-88), is in its first issue and will probably be modified. It is therefore important that the precise details of the algorithms used to calculate K_g as well as the validity criteria be documented. If crack-arrest methodology does change, that data can be re-evaluated using the new methodology. 10 | THE NAN," OF THIS PROGRAM IS CA_TEST 20 1 30 DIM Banner\$[40] 40 Banner\$ = "Version 5.1 Revised February 14, 1990)" 50 60 70 1 THIS PROCRAM PERFORMS CRACK ARREST TEST COMPUTATIONS 80 1 90 100 INTEGER Cycle, Nrec, Nhytes, Nvar 120 1 130 1 INITIALIZATION OF VARIABLES 140 1 150 1 Data stored = 0 (FALSE) (DEFAULT) data for specimen .ave NOT been stored = 1 (TRUE) data for this spec. have been stored previously 160 1 170 1 This is set in subprogram Load after succesful input 180 1 190 | For DUPLEX specimens, Ty = measured value, and for 200 | WELD-EMBRITTLED specimens Ty = 0 210 1 220 1 Yao = Dynamic yield strength increment is set at 205 MPa per ASTM 1221-88 Para. 230 1 240 1 250 ON ERROR GOTO Error sub 260 1 270 Yso=205 280 Data stored=0 290 Msus\$ = ".. 700.1" 300 PRINTER IS 1 310 1 320 PRINT 'CRACK ARREST CALCULATIONS '& Banners' 330 PRINT "THIS PROGRAM LOADS DATA PREVIOUSLY STORED ON 3.5" DISKS" 340! WAIT 2 350 1 360 1 370 1 seessessesses Bogin seeses 380 Begin: ! 390 MS="Y" 400 INPUT "LOAD DATA FROM DISKETTE? (Y/N) Y".MS 410 IF MS = "Y" THEN 420 PRINTER IS 1 430 DISP "INSERT DATA SOURCE AND PRESS CONTINUE" 440 PAUSE 450 GOSUB Load 460 GOTO Print dta 470 END IF 480 1 490 1 ******** Start ******* 500 1 510 1 Starts data input from Keyboard 520 Start: 530 PRINTER IS 1 540 PRINT USING "@" 550 INPUT ' ENTER SPECIMEN NO. ', File\$ 560 PRINT USING ***SPECIMEN NO. = **.10A* File\$ 570 PRINT 580 INPUT 'ENTER TEST TEMPERATURE IN C'.Tp 590 INPUT 'ENTER YIELD STRENGTH AT TEST TEMPERATURE IN MP#' Ys 600 INPUT "ENTER TEST DATE IN THE FORM 072989 (6 CHARACTERS MAX)". Test_date

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610 PRINT USING *3X.**TEST DATE =**.1X.6Z*.Test_date
620 PRINT USING ***YS = **.10D.2X.9A.5D.4A *:Ys. MPs at Tp. C*
630 INPUT 'ENTER E IN MP#'.E
640 PRINT USING ***E = **.10D .2X.4A*:E.*MPa*
6.30 PRINT
660 INPUT *ENTER B IN INCHES*.B
670 FRINT USING ***B = **,3D.3D.2X.4A*.B.*inch*
680 INPUT "ENTER EN IN INCHES".Bn
690 PRINT USING ***BN = 0.3D.2X.4A* Bn. "inch"
700 INPUT *ENTER 2H IN
                          .......H
710 PRINT USING **
                         3D.2X.4A*11.*inch*
                        - .65*.Wt
720 INPUT *ENTER W
730 PRINT USING ***W
                            3D.3D.2X.4A*;Wt.*inch*
740 PRINT
750 INPUT "ENTER PR IN mm", Pr
760 PRINT USING *** PR = **.3D.2D.3X.4A*.Pr.*mm *
770 INPUT 'ENTER RS IN mm', Rs
780 PRINT USING ***RS = **.3D.2D.3X.4A*.Rs.*mm *
790 INPUT 'ENTER RT IN mm", Rt
800 PRINT USING ***RT = **.3D.2D.3X.4A*.Rt.*mm *
810 INPUT 'ENTER TU IN mm', Tu
820 PRINT USING ***TU = **.3D.2D.3X.4A*:Tu.*mm *
830 PRINT
840 INPUT *ENTER N IN mm*.N
850 PRINT USING ***N = **.3D.2D.3X.4A*:N.*mm *
860 1
870 INPUT 'ENTER TV IN mm. (= 0 for WELD-EMBRITTLED !!!)', Ty
$80 PRINT USING */,K.3D.3D.2X.2A*, "TV = ".Tv."mm"
890 1
900 1
                                 **** Correc () ****
910 Cor 8c 0: 1
920 CL $ ... Y"
930 IN 'UT "CORRECTIONS? (Y/N) Y", Cods
940 IF JPC$(Cor$)="Y" THEN GOTO Start
950 1
960 1
970 Ston $ = "Y"
980 : NP /T *STORE DATA ON DISKETTE? (Y/N) Y* Store$
990 IF L PC$(Store$) = 'Y' THEN GOSUB Store
1000 1
1010 1
1030 1
                               ***** Print dia ******
1040 Print_dta: 1
1050 1 Prints input data if required
1060 GOSUB Hardcopy
1070 !
1080 1
1090 1 READ LABEL Label$ FROM Msus$
1100 1
1120 GOSUB Specimen
1130 1
1140
1150 PRINT USING */.9X.**YIELD STRENGTH = **.5D.1X.JA.#*.Yx.*MP#*
1160 PRINT USING '8X." YOUNG'S MODULUS = "."D.IX.3A.#" E. MP#"
1170 PRINT USING *** INCR. TO YIELD (DYN) (Yso) #** 4D.1X.3A*; Yso, 'MP#*
1180 1
1190 1
```

1200 PRINT USING */,**Pre-test specimen dimension measurements***

1210 ! 1220 3 1230 PRINT USING */ 9X.**B * ** 3D.3D.2X.4A* B.*inch* 1240 PRINT USING *9X,**BN = **,3D.3D.2X.4A*;Bn.*inch* 1250 PRINT USING '9X, **2H = **, 3D, 3D, 2X, 4A*; H, *inch* 1260 PRINT USING '9X, **WT = **, 3D, 3D, 2X, 4A*; WL*inch* 1270 PRINT 1280 PRINT USING "9X,""PR # ".3D.2D.2X.4A",Pr."mm" 1290 PRINT USING '9X. **R5 = **,3D.2D.2X.4A*.Rs. mm* 1300 PRINT USING '9X, "'RT = ".3D.2D.2X.4A", RL, "mm" 1310 PRINT USING '9X." TU + "".3D.2D.2X 4A". Tu. "mm" 1320 PRINT USING "9X.""N = "",3D.2D.2X.7A.D.6D.2A";N."mm" 1330 1340 PRINT USING */.9X.K.3D.2D.2X.2A*; "TV = ".Tv."mm" 1350 1360 1370 PRINT USING */,**Fre-test specimen dimension calculations*** 1380 1 1390 W = W(*25.4-(Rs+R1)/2 1400 Ao=Tu+(RI-Rs)/2 1410 Ro= An/W 1420 1430 PRINT USING */.9X,**W = "*,4X,6D.2D,2X,2A";W,"mm" 1440 PRINT USING "9X." A0 = "4X.6D.2D.2X.2A": A0. "mm" 1450 PRINT USING *9X.**Aa/w = **.4X.D.3D*.Ro 1460 1470 1480 Faow=(2.24*(1.72-9*Ro+Ro*2)*(1-Ro)*.5)/(9.85-17*Ro+11*Ro*2) 1490 PRINT USING "9X, "*F(Ao/W) = ".4X, D.4D"; Faow 1500 1 1510 Ff=Pr+(Rt+Rs)/2 1520 Floverw = FUW 1530 PRINT USING "/.9X." Clip gage is located at (FF/W) = ".4X.D.2D" Floverw 1540 1550 Flow cor#.25 1560 1F (Ffoverw > 98*Ffow_cor) AND (Ffoverw < 1.02*Ffow_cor) THEN 1570 GOTO Proceed 1580 ELSE 1590 PRINT *WARNING !!! Do not proceed to test this specimen* PRINT "Check input data, particularly PR, RS, and RT" 1600 PRINT "Clip gage location blocks are NOT located at FF/W = 0.25 !" 1610 DISP "INCORRECT (FF/W)| IF POSSIBLE CORRECT DATA BEFORE PROCEEDING" 1620 Ques = "Y" 1630 1640 INPUT "DO YOU WANT TO CORRECT INPUT DATA? (Y/N) Y'.Ques 1650 IF UPC\$(Que\$) = 'Y' THEN GOTO Sum 1660 Ques = "Y" 1670 INPUT "Do you want to QUIT now ? (Y/N) Y", Que\$ 1680 IF UPC\$(Que\$) = 'Y' THEN GOTO Finish 1690 END IF 1700 1 1710 1 1720 Proceed: 1 1730 1F Tv < >0 THEN 1740 Ebcl = (R(-Rs)/2 + Tv 1750 PRINT USING */.9X.**EBCL # ** 10X.4D.2D.4A* Ebcl.* mm* 1760 END IF 1770 1 1780 1790 PRINT USING */,K,/*, *Pre-test loading calculations* 1800 1

```
1810 1 Fact is in mils. W is in mm
1820 Fact=1.E+3*(Ys*W/25.4*(Bn/B)*.5)/(E*Faow)
1830 Deltacinit= 69*Fact
1840 1
1E50 Num cycles = 10
1860 IF Tv < >0 THEN
1870 Num cycles == 1
1880 PRINT USING */.K./*: "This is a DUPLEX specimen, only one loading cycle
tipeded*
1890 END IF
1900 FOR Cycle=1 TO Num cycles
1910 Dol max = (1 + .25*(Cycle-1))*Deltaoinit
1920 !
1930 1100 mils = 10 volts
1940 Voluse.1*Del max
1950 PRINT USING *11X,**FOR CYCLE No. =**,5D,**. maximum COD =**,5D,D,** m
ils**,15D.2D,** Volts***;Cycle,Del max.Volts
1960 NEXT Cycle
1970 Deltaomax = 1.5*Fact
1980
1990 IF Tv=0 THEN PRINT USING */.11X,**Probable maximum COD for useful results*
*.5D.D.** Mils***;Deltaomsx
2000 1
2010 Cor$ = "Y"
2020 INPUT 'DO YOU WANT TO MAKE CORRECTIONS IN THE STORED DATA? (Y/N) Y* Curs
2030 IF Cor$ = "Y" THEN GOTO Start
2040 1
2050 Prints = "Y"
2060 INPUT 'DO YOU WANT TO PRINT OUT DATA (SCREEN/PRINTER) ? (Y/N) Y* Prints
2070 IF UPC$(Print$)="Y" THEN GOTO Print dia
2030 1
2090 Ptc$ = "Y"
2100 INPUT 'DO YOU WANT TO DO POST TEST CALCULATIONS? (Y/N) Y', Pics
2110 IF UPC$(Ptc$)="Y" THEN Post teale
2120 GOSUB Hardcopy
2130 Que$="Y"
2140 INPUT "DO YOU WANT TO DO MORE PRE TEST CALCULATIONS? (Y/N) Y". OwS
2150 IF UPC$(Que$)="Y* THEN Begin
2160 Que$ = "N"
2170 INPUT "DO YOU WANT TO STORE THE DATA (Y/N) N* . Ques
2180 IF UPC$(Que$) = "Y" THEN GOSUB Store
2/90 Ques = "Y"
2200 INPUT 'DO YOU WANT TO DO MORE? (Y/N) Y', Oues
2210 IF UPC$(Que$) = "Y" THEN GOTO Begin
2220 GOTO Finish
2230 STOP
2240 1
2250
2260 1
                                ****** Post tcalc
                                                 ....
2270 Post teale: 1
2280 PRINTER IS I
2290 ! Check if only pre-test data was stored, but no post-test data
2300 Psisdia$ = "Y"
2310 INPUT 'HAS POST TEST DATA FOR THIS TEST BEEN PREVIOUSLY STORED! (Y/N) Y*P
stschaS
2320 IF UPC$(Pstsdta$) = "Y" THEN
2330 1
2340 1 CODs have been stored in inches. Initialize variables R1, R3, P4 & P5
2350 ! for printing out in mils.
2360 1
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```
2370 R1= Deltapsb1*1 E+3
2400 P5 = Doltan*1.E+3
2410 GOTO Pr dia2
2420 END IF
2430 1
2440 1
                                 seas Correc | seas
2450 1 Pst-test data input from keyboard
2460 Correc 1: 1
2470 INPUT 'ENTER W-As AT + BN/4 IN mm', Waat
2480 PRINT USING *** W-As AT # BN/4 =** 4X.3D.2D.2X.4A*:Wast.*mm *
2490 INPUT 'ENTER W-Aa AT MIDLINE IN mm'. Waam
2500 PRINT USING *** W-As AT MIDLINE =** 4X,3D.2D.2X,4A* Wsam, "mm
2510 INPUT 'ENTER W-As AT -BN/4 IN mm', Waab
2520 PRINT USING *** W-A# AT -BN/4 =**.4X.3D.2D.2X.4A*.Wash.*mm *
2530 PRINT
2540 PRINT
2550 INPUT 'ENTER RI IN mils', RI
2560 PRINT USING ***R1 = **.2X.4D.D.** mds*** R1
2570 INPUT 'ENTER R3 IN mils', R3
2580 PRINT USING ***R3 =**.2X,4D.D.** mils***:R3
2590 INPUT 'ENTER P4 IN mils' .P4
2600 PRINT USING *** P4 = **, 2X, 4D. D. ** mils***, P4
2610 INPUT "ENTER P5 IN milk", P5
2620 PRINT USING ***P5 =**,2X,4D,D,** muls***;P5
2630 PRINT
2640 INPUT "ENTER Pmax IN pounds", Pmax
2650 PRINT USING *** Pmax =**,4X,6D,2X,4A*;Pmax,*lbs *
2660 INPUT *ENTER Pmin IN pounds*, Pmin
2670 PRINT USING *** Prain
                                = **,4X,6D,2X,4A*;Pmin,*ibs *
2680 Cor$ = "Y"
2690 INPUT "CORRECTIONS" (Y/N) Y' Cors
2700 IF Cor$ = "Y" THEN
2710 PRINT USING '@'
2720 GOTO Correc_1
2730 END 15
2740 1
2750 1
2760.1 For compatibility with previous data stored on diskettes, CODs
2770 1 are stored in inches
2780 Deltapsb1=R1*1.E-3
2790 Sdultap = R3*1.E-3
2800 Deltao = P4+1.E-3
2810 Deltaa=P5*1.E-3
2820 1
2830 1
2840 Store$ = "Y"
2850 INPUT *STORE DATA ON DISKETTE! (Y/N) Y*, Stores
2860 IF Store$ = "Y" THEN GOSUB Store
2870
2880 1
                                   comme pr dia2 eree
2890 Pr_dta2: 1
2900 GOSUB Hardcopy
2910 1
2920 GOSUB Specimen
2930 !
2940 1
2950 PRINT USING */, ** Post-test crack arrest measurements***
2960 PRINT USING */.9X,**W-Aa AT +BN/4 =**,4X,3D.3D,** num***,Waat
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2970 PRINT USING *9X,**W-As AT MIDLINE = **.4X.3D.3D.** mm***;Waam 2980 PRINT USING *9X, **W-As AT -BN/4 = **,4X,3D,3D,** mm***; Wash 2000 1 3000 1 3010 PRINT USING */.9X .**Zaro load disp. offset at end of cycle 1 (R1) +* ".4D.D."" mils""".R1 3020 PRINT USING *9X, **Total disp. offset at end of cycle n-1 (R3) = 15. 4D.D.** mils***;R3 3030 PRINT USING *9X.**Diaplacement at onset of unstable cruck growth (P4) = **. 4D.D.** mils***;P4 3040 PRINT USING *9X.**Displacement approx ... I s after arrest (P5) = **. 4D.D.** mils *** P5 3050 1 3060 1 3070 FRINT USING */.9X,**Pmax = **.6D.2X.3A*.Pmex.*Ihs* 3080 PRINT USING * 9X, **Pmin = **,6D,2X,3A*,Pmin.*Ibs * 3090 Rp=(Pmax-Pmin)/Pmax 3100 PRINT USING "9X." (Pmax-Pmin)/Pmex =" .5D.2D"; Rp 3110 IF Rp > 5 THEN 3120 PRINT USING *9X,**Load drop guideline met*** 3130 ELSE 3150 END IF 3160 1 3170 3180 3190 PRINT USING */.** Post-test crack arrest calculations*** 3200 1 3210 W_as_svg=(Wasm+Waat+Waab)/3 3220 PRINT USING */.9X,**Length of remaining ligament W-Aa(average) = **.4D.D.** mm***;W_as_evg 3230 1 3240 1 Aa = arrested crack length per ASTM Paragr. 8.6 3250 A#=W-W_88_8V8 3260 Crakimp=Aa-Ao 3270 PRINT USINO */.9X.**Crack jump Aa - Au**.24X.** =**.4D.D.1X.2A*.Crakimp.*m m 3280 1 3290 1 3300 1 Do and Da are "net" CODs per ASTM (in m) 3310 Do=(Deltao-Sdeltap)*25.4 1.E-3 3320 1 For duplex specimens, there is only one unload cycle 3330 1 \$ Deltapsb1 (R1) is equal to Sdeltap (R3) 3340 1 3350 Da= .5*(Deltao + Deltas-Deltapsh1-Sdeltap)*25.4*1.E-3 1360 R/= A&/W 3370 Fafw = (2.24*(1.72-9*Rf+Rf*2)*(1-Rf)*.5)/(9.85-17*kf+11*Kf*2) 3386 Fac=E*(B/(Bn*W*.001))*.5 3390 Ko=Do*Faow*Fac 3400 Kf=Da*Fafa*Fac 3410 1 3420 1 3430 PRINT USING */.9X. "Net COD at initiation per ASTM = "*.2X.1D.3DE." mi***:D 3440 PRINT USING "9X.""Net COD at arrest per ASTM = ".2X.1D.3DE." m""".Da 3450 -1 3460 ! 3470 PRINT USINO */,9X." Arrested crack length (Aa) w**.5D.D.** mm ***;A4

3480 PRINT USING "9X."*Fractional arrested crack length (Au/w) = ".4X.D.3D":Rf

3490 PRINT USING *9X.**Geometry factor F(Aa/W) 3500 3510 3530 PRINT USING '9X, '** '', ''Ko = '', 5D, *' MPa.m'0.5'', ** ***' Ko 3540 PRINT USING '9X, *** '', ''Ka = '', 5D, *' MPa.m'0.5'', ***'' Ki 3566 3570 VALIDITY CRITERIA 3580 3590 3600 1 3610 W015=.15*W 3620 Liga b=1.25*10.0*(Kf/(Ys+Yso))*2 3630 Liga_c=1000*/Kf/(Ys+Yso))*2 3/540 Nvalid=2*N 3630 Minimp=10LD*(1.o/Ys)*2/(2*P1) 3660 1 3670 ! 3680 PRINT USING 1/1X, 11 ALIDITY CRITERIA PER ASTM 1221-88 PARA 9.3*** 3690 PRINT USING *17,40X,**ACTUAL VALUES**,40X,**ASTM CRITERIA MINIMUMS*** 3700 37:0 RIGAR CARADRANA CARABAR CARABARA 3720 ! * REMAINING LIGAMENT CRITERIA "A" & "B" PER ASTM * 3730 1 3740 1 3750 PRINT USING */,1X,**Length of remaining ligament (W - Aa) - - ,4D,D,1X,4A,# ";W as avg,"mm" 5760 PRINT USING *1, X, ** Remaining rigament (0.15W)**, 17X, ** = ** AD, D, 1X, 2A*; W015 *mm* 3770 IF W_ma_evg > W015 THEN PRINT USING "IX.K", "REMAINING LIGAMENT CRITERION "7." MET" 3780 3790 ELSE 3800 PRINT ****REMAINING LIGAMENT CRITECION ** A** NOT MET**** 3810 END IF 3820 3830 PRINT USING */.1X.**Length of remaining ligament (W - Aa) =**.4D.D.1X.4A.# ';W_aa_avg, 'mm' 3840 PRINT USING *15X, ** Remaining ligament 1.25* [Ka/(YS+YSo)]*2 =**, 4D.D, 1X, 4A";Liga_b,"mm" 3850 IF W as avg > Liga b THEN 3860 PRINT USING "IX.K". "REMAINING LIGAMENT CRITERION "B" MET" 3870 ELSE 3880 PRINT ****REMAINING LIGAMENT CRITERION **B** NOT MET**** 3890 END IF 1900 3910 3920 ! 3930 1 recordecees hereeceeseesee 3940 1 e THICKNESS CRITERION (C) e 3550 1 3960 1 3970 Bmm=B*25.4 3980 PRINT USING */,1X .** Spectation width B =**,3D.3D,** inches**.6X,**=**,4D.D.* " nam",#":B.Bmm 3990 PRINT USING *17X."*Specimen width [Kx/(YS+YSo)]*2**,12X.**=**.4D.D.** mm* "":Ligs_c 4000 IF Binia > Liga_c THEN

1

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4010 PRINT USING *11X.K*; 'THICKNESS CRITERION **C** MET*

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4020 ELSE
4030 PRINT ****THICKNESS CRITERION **C** NOT MET****
4040 END IF
4050 1
4060 !
4070
              udddddddddddddddddddddddddddddddddddd
4080 1
4069.1
              d Crack jump length criteria (D) and (E) d
4100 1
             4110 1
4120 1
4130 1
4140 1
4,50 1
4160 IF TV×0 THEN
4170 PRINT USING */,1X,**Crack jump Aa - Ao**,20X,** =**,1X,3D,D,1X,2A,#*(Cra
kimp."mm"
     PRINT USING *17X, ** ASTM 2N**,36X, ** = **,3D.D.1X,2A*; Nvali, *mm*
4180
4190 IF Crakjmp > Nvalid THEN
       PRINT USING *11X, K*, *WELD EMBRITTLED SPLC. CRACK JUMP-LENGTH CRITERI
4200
ON **D** MET*
4210 ELSE
      PRINT ****WELD EMBRITTLED SPEC. CRACK JUMF-LENGTH CRITERION **D** NO
4220
T MET****
4230 END IF
4240 1
4250 1
4260 PRINT USING */.1X.**Crack jump Aa - Ao**.20X.** = **.1X.3D.D.1X.zA.5*:Cra
kimp, "mm"
4270 PRINT USING *17X.**Min. crack jump (Ko/YS)*2/2PI**, I2X.** --**.3D.D.IX.
2A*:Minimp, "mm"
4280 IF Crakimp > Minimp THEN
       PRINT USING "IX.K", "WELD EMBRITTLED SPEC. CRACK JUMP LENG TI CRITERIO
4290
N ""E" MET'
4300 ELSE
       PRINT ****CRACK JUMP LENGTH CRITERION **L** NOT MET****
4310
4320 END IF
4330 1
                 ! DUPLEX SPECIMEN CRITERION 'D' + 'E'
4340 ELSE
4350 1
               ("D" & "E" ARE REPLACED WITH A SINGLE ONE)
4360 !
4370 Dpx/mp=Aa-Ebcl
4340 PRINT USING */.1X.**Crack jump beyond EB weld (Aa - EBCL) = **.1X.3D D.2
X.4A.#":Dpsjmp,"mm"
4390 1
 4400 !
 4410
      Enmm=Bn*25.4
 4420 PRINT USING *14X,**Specimen width at notch (Bn) =**,D.3D,** inches = **
 .JD.D."" mm"""Bn.Bnmm
 4430 IF Dpxjmp > Bnmm THEN
       PRINT *DUPLEX SPECIMEN CRACK JUMP CRITERION **D** + **E** MET*
 4440
 4450
       ELSE
        PRINT ****DUPLEX SPEC. CRACK JUMP CRITERION **D** + **E** NOT MET***
 4460
 4470 END IF
 4480 END 1F
 4490 1 END OF CRITERIA 'D' & 'E' DETERMINATIONS
 4500 1
 4510 1
 4520 Cor$="Y"
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C

4530 INPUT 'DO YOU WANT TO CORRECT POST TEST DATA (Y/N) Y'.Cor\$ 4540 IF UPC\$(Cor\$) = "Y" THEN 4550 PRINTER 15 1 4560 GOTO Correc, 1 4570 END IF 4580 Ques\$ = "Y" 4590 INPUT "DO YOU WANT TO DO MORE (Y/N) Y".Quess 4600 IF UPC\$(Ques\$)= "Y" THEN Begin 4610 GOTO Finish 4620 1 4530 1 4640 1 ******* Store ******** 4650 Store: ! 4660 ION ERROR GOSUB Error 4670 DISP *INSERT INITIALIZED AND LABELLED DISC IN DRIVE AND PRESS CONTINUE: 4680 PAUSE 4690 ! 4700 READ LABEL Label\$ FROM Msus\$ 4710 1 4720 4730 ! Data stored = 1 (TRUE) if data has been previously stored. 4750 IF Data_stored THEN PURGE File\$&Msus\$ 4790 4800 ! There are 26 variables. Round up to 30 4810 Nrec = 1 4820 Nvar=30 4830 Nbytes = 8*Nvar 4840 CREATE BDAT File\$&Msus\$.Nrec.Nbytes 4850 ASSIGN @Path TO File\$&Msus\$;FORMAT OFF 4860 ! 4870 ! Note: Breet, Himet, Witnet, and Win are not used, but for compatability 4880 1 with previously written output, its location is reserved. 4890 ! It can be used for other purposes. 4900 1 Following line is one used to date (July 8, 1989) 4910 1 OUTPUT @Path;File\$,Ys,Tp,E,B,Brret,Bn,H,Hmet,Wt,Wtmet,Wn,Pr,Rs,Rt,Tu,Tv,N Waam, Waat, Waab, Deltapsh1, Sdeltap, Deltao, Deltaa, Pmax, Pmin 4920 OUTPUT @Path;File\$,Ys,Tp,E,B,Test_date,Bn,H,Hmet,Wt,Wtmet,Wn,Pr,Rs,Rt,Tu,T v, N, Waam, Waat, Waab, Deltapsb1, Sdeitap, Deltao, Deltaa, Pmax, Pmin 4930 ASSIGN @Path TO * 4940 Data_stored = 1 4950 DISP *DATA STORAGE SUCCESSFUL* 4960 WAIT 1 4970 DISP * * 4980 RETURN 4990 ! 5000 1 sessesses Load sesses 5010 Load:1 5020 1 5030 5040 INPUT *SPECIMEN NO. *, File\$ 5050 READ LABEL Label\$ FROM Msus\$ 5060 1 5070 ASSIGN @Path1 TO File\$&Mses\$;FORMAT OFF 5080 ! Bmet, Hmet, Witnet, & Wn is no longer used, but location is kept for com patability

5090 ! See comment in Store routine

5100 READ LABEL Label\$ FROM Maus\$ 5110 ENTER @Path1:File\$.Ys.Tp.E.B.Test_date.Bn.H.)Imet.Wt.Wtmet.Wn.Pr.Rs.Rt.Tu.T v. N. Waam, Waat, Waab, Deltapsb1, Sdeltap, Deltao, Deltas, Pmax, Pmin 5120 ASSIGN @Path1 TO * \$130 11 data input from storage was successful, then data is actually 5140 1 there Initialize Data stored = 1 (TRUE) 5150 Data stored = 1 5160 DISP "DATA LOA" FROM DISK "& Msusta" "& Label\$&" SUCCESSFUL" 5170 WALC3 5180 RETURN 5190 1 5200 1 ***** Advance ***** 5210 Advance: ! 5220 PS = "N" 5230 INPUT 'ADVANCE PAPER? (Y/N) N*, P\$ 5240 IF UPC\$(P\$) = "Y" THEN PR' 'IT USING "@" 5250 RETURN 5260 1 5270 1 5280 1 ***** Hardcony ***** \$290 Hardcopy: 1 5300 HS="N" 5310 INPUT *NEED HARDCOPY? (Y/N) N* HS 5320 IF UPC\$(H\$) = "Y" THEN 5330 PRINTER IS 9 5340 GOSUB Advance 5350 PRINT USING *K,2X,K,3X,K,2X,K*,*CRACK ARREST CON . JTER CODE CA TEST *& Ba nner\$, "Time of this run: ".DATE\$(TIMEDATE), TIME\$(TIMEDATE) 5360 ELSE 5370 PRINTER IS 1 5380 END IF 5390 RETURN 5400 1 5410 ! **** Error sub **** 5420 Error sub: 1 5430 PRINT ERRM\$ 5440 PRINT "POSSIBLE INPUT DATA ERROR: CHECK INPUT" 5450 DISP "HIT CONTINUE" 5460 PAUSE 5470 GOTO Begin 5480 1 5490 1 **** Specimen **** 5500 Specimen: 1 5510 PRINT USING *2/.1X.**SPECIMEN NO. **.7A.#*.File\$ 5520 IF Tv < >0 THEN 5530 PRINT USING *** (DUPLEX)**.#* 5540 ELSE 5550 PRINT USING *** (WELD EMBRITTLED)**.#* 5560 END IF 5570 ! 5580 ! 5590 PRINT USING '9X, "*TEST TEMPERATURE = "*,4D,1X,A,#",Tp,*C* 5600 PRINT USING "3X,K,#";" from disk "&Label\$ 5610 PRINT USING "3X.""TEST DATE = "". IX.6Z":Test date 562ⁿ RETURN 5630 1 5640 ! **** Dox we **** 5650 | Dpx_we: | This subprogam prints out whether specimen 5660 1 is a duplex or weld-embrittled type and value of Tv \$670 1 Note that Tv is set equal to zero for weld-embrittled

5680 ! specimens, and is the measured value for duplex specimens 5690! IF Dpaspm\$ = "Y" THEN 5700! PRINT USING "/, ""This is a DUPLEX specimen"" 5710! ELSE 57201 PRINT USING */,K*,*This is a WELD EMBRITTLED specimen and TV is set = 0 by program" 5730! END IF 5740! PRINT USING */,9X,K,3D,2D,2X,2A*;*Tv = *,Tv,*mm* 5750! RETURN 5760 1 5770 1 **** Finish **** 5780 Finish: 1 End of Program 5790 PRINTER IS 1 5800 DISP *Crack arrest program ended.* 5810 STOP 3820 END

AFPENDIX C

YOUNG'S MODULUS USED IN THE EVALUATION OF THE CRACK ARREST DATA Young's modulus is used to calculate the stress intensity factors K_o and K_a , both of which are directly proportional to the value of E used [see Eq. (C.1)]. Young's modulus together with the yield strength are used to calculate the load increment for the loading and unloading cycles for weld-embrittled type specimens as well as the validity criteria.

The Young's modulus used to evaluate both unirradiated and irradiated crack arrest data is calculated from the following expression [1]:

$$E = 207.2 - 0.0571T$$
, (C.1)

where E = Young's modulus in GPa, and T = temperature in °C. Reference [1] in turn cites Subsection NB 2300, ASME Boiler and Pressure Vessel Code, Sect. III. The current version of NB 2300 does not give this equation, so it is of interest to compare it with values in the recent edition of the ASME Code [2]. Figure C.1 and Table C.1 compare Young's modulus calculated at the tabular temperatures given in the American Society of Mechanical Engineers Code. In the temperature range used for testing both the unirradiated and irradiated crack arrest specimens, -75 to 100°C, Eq. (C.1) gives values that are about 2% higher than those in Ref. [2]. It is planned to measure Young's modulus for the weldments and reasonable the crack-arrest data.

Temperature	Young's m	% Difference			
	ASME (A)	EPRI (E)	[(E+A)/A		
-198	216.5	218.5	1		
-129	212.4	214.6	ĩ		
-73	208.2	211.4	2		
21	203.4	206.0	o na sina a		
93	198.6	201.9	2		
149	195.1	198.7	2		
204	191.0	195.5	2		
260	188.2	192.4	2		
316	184.1	189.2	3		
371	175.8	186.0	6		
427	166.9	182.9	10		

Table C.1. Comparison of Young's modulus calculated using values given in the ASME Code and Eq. (C.1)

NOTE :

ASME = American Society of Mechanical Engineers EPRI = Electric Power Research Institute

REFERENCES FOR APPENDIX C

- W. L. Server, J. W. Sheckherd, and R. A. Wullaert, Electric Power Research Institute, Falo Alto, Calif., Fracture Toughness Data for Ferritic Nuclear Pressure Vessel Materials, EPRI NP-119, April 1976.
- ASME Boiler and Pressure Vessel Code, Sect. III, Div. 1, Appendix I, Table I-6.0, American Society of Mechanical Engineers, July 1, 1989.

APPENDIX D

TYPICAL OUTPUT FROM CA_TEST FOR A WELD-EMBRITTLED AND DUPLEX SPECIMEN

CRACK ARREST COMPUTER CODE CA_TEST Version 5.1 Revised February 14, 1990) Time in this retail 12 Sep 1990 16:02:48

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Pre-test specimen dimension measurements

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Pre-test specimen dimension calculations

Clip gage is located at (FF/W) = .25

Pre-test loading calculations

FOR CYCLE No. = 2, maximum COD = 38.1 mils 3.81 Volt FOR CYCLE No. = 3, maximum COD = 45.8 mils 4.58 Volt FOR CYCLE No. = 4, maximum COD = 53.4 mils 5.34 Volt FOR CYCLE No. = 4, maximum COD = 51.4 mils 5.34 Volt FOR CYCLE No. = 5, maximum COD = 61.0 mils 6.10 Volt FOR CYCLE No. = 6, maximum COD = 68.6 mils 6.86 Volt FOR CYCLE No. = 7, maximum COD = 76.3 mils 7.63 Volt FOR CYCLE No. = 8, maximum COD = 83.9 mils 8.39 Volt FOR CYCLE No. = 9, maximum COD = 91.5 mils 9.15 Volt FOR CYCLE No. = 10, maximum COD = 92.1 mils 9.91 Volt	FOR	CYCLE	No.	90	1.1	maximum COD =	30.5	mils	3.05 Volts
FOR CYCLE No. 3. maximum COD 45.8 mils 4.58 Volt FOR CYCLE No. 4. maximum COD 53.4 mils 5.34 Volt FOR CYCLE No. 5. maximum COD 61.0 mils 6.10 Volt FOR CYCLE No. 5. maximum COD 68.6 mils 6.86 Volt FOR CYCLE No. 6. maximum COD 76.3 mils 7.63 Volt FOR CYCLE No. 7. maximum COD 83.9 mils 8.39 Volt FOR CYCLE No. 8. maximum COD 91.5 mils 9.15 Volt FOR CYCLE No. 9. maximum COD 99.1 mils 9.91 Volt	FOR	CYCLE	No.	36	2.	maximum COD =	38.1	mils	3.81 Volts
FOR CYCLE No. = 4, maximum COD = 53.4 mils 5.34 Volt FOR CYCLE No. = 5, maximum COD = 61.0 mils 6.10 Volt FOR CYCLE No. = 5, maximum COD = 68.6 mils 6.86 Volt FOR CYCLE No. = 7, maximum COD = 76.3 mils 7.63 Volt FOR CYCLE No. = 7, maximum COD = 83.9 mils 8.39 Volt FOR CYCLE No. = 9, maximum COD = 91.5 mils 9.15 Volt FOR CYCLE No. = 10, maximum COD = 99.1 mils 9.91 Volt	FOR	CYCLE	No.		3.	maximum COD =	45.8	mils	4.58 Volts
FOR CYCLE No. 5. maximum COD 61.0 mils 6.10 Volt FOR CYCLE No. 6. maximum COD 68.6 mils 6.86 Volt FOR CYCLE No. 7. maximum COD 76.3 mils 7.63 Volt FOR CYCLE No. 8. maximum COD 83.9 mils 8.39 Volt FOR CYCLE No. 9. maximum COD 91.5 mils 9.15 Volt FOR CYCLE No. 10. maximum COD 99.1 mils 9.91 Volt	FOR	CYCLE	No.	88	4,	maximum COD =	53.4	mils	5.34 Volts
FOR CYCLE No. 6. maximum COD 68.6 mils 6.86 Volt FOR CYCLE No. 7. maximum COD 76.3 mils 7.63 Volt FOR CYCLE No. 8. maximum COD 83.9 mils 8.39 Volt FOR CYCLE No. 9. maximum COD 91.5 mils 9.15 Volt FOR CYCLE No. 10. maximum COD 99.1 mils 9.91 Volt	FOR	CYCLE	No.	*	5.	maximum COD =	61.0	mils	6.10 Volts
FOR CYCLE No. = 7. maximum COD = 76.3 mils 7.63 Volt FOR CYCLE No. = 8. maximum COD = 83.9 mils 8.39 Volt FOR CYCLE No. = 9. maximum COD = 91.5 mils 9.15 Volt FOR CYCLE No. = 10. maximum COD = 99.1 mils 9.91 Volt	FOR	CYCLE	No.		6.	maximum COD =	68.6	mals	6.86 Volts
FOR CYCLE No. 8, maximum COD 83.9 mils 8.39 Volt FOR CYCLE No. 9, maximum COD 91.5 mils 9.15 Volt FOR CYCLE No. 10, maximum COD 99.1 mils 9.91 Volt	FOR	CYCLE	No.	82	7.	maximum COD =	76.3	mils	7.63 Volts
FOR CYCLE No. 9, maximum COD 91.5 mils 9.15 Volt FOR CYCLE No. = 10, maximum COD = 99.1 mils 9.91 Volt	FOR	CYCLE	No.	30	8,	maximum COD =	83.9	mils	8.39 Volts
FOR CYCLE No. = 10, maximum COD = 99.1 mils 9.91 Vol	FOR	CYCLE	No.	-	9,	maximum COD =	91.5	mils	9.15 Volts
	FOR	CYCLE	No.	10	10,	maximum COD :	= 99.1	mils	9.91 Volts

Probable maximum COD for useful results 66.3 Mils

CRAC's ARREST COMPUTER CODE CA_TEST Version 5.1 Revised February 14, 1990) Time of this run: 12 Sep 1990-16:03:09

JPECIMEN NO. A72W48 (WELD EMBRITTLED) TEST TEMPERATURE = -30 C from disk CANO1 TEST DATE = -0000

ASTM CRITERIA MINIMUMS

Post-test crack arrest measurements

W-Aa AT + BN/4 = 35.192 mm W-Aa AT MIDLINE = 34.573 mm. W-Aa AT -BN/4 = 33.037 mm

Zero load disp. offset at end of cycle 1 (R1) = 1.5 mils Total disp. offset at end of cycle n-1 (R3) = 3.0 mils Displacement at onset of unstable crack growth (P4) = 47.8 mils Displacement approx. .1 s after arrest (P5) = 51.4 mils

Pmax = 12000 lbs Fmin = 750 lbs (Pmax-Pmin)/Pmax = .94 Load drop guideline met

Post-test crack arrest calculations

Length of remaining ligament W-Aa(average) = 34.3 mm

Crack jump As - Ao = 49.2 mm

Net COD at initiation per ASTM = 1.138E-03 m Net COD at arrest per ASTM = 1.203E-03 m

Arrested crack length (Aa) = 92.5 mm Fractional arrested crack length (Aa/w) = .730 Geometry factor F(Aa/W) = .1193

* Ko = 194 MPa.m^{*}0.5 * * Ka = 98 MPa.m^{*}0.5 * **********************

VALIDITY CRITERIA PER ASTM 1221-88, PARA, 9.3

ACTUAL VALUES

Length of remaining ligament (W - Aa) = 34.3 mm REMAINING LIGAMENT CRITERION *A* MET	Remaining ligament (0.15W) = 19.0 mm
Length of remaining ligament (W - Aa) = 34.3 mm REMAINING LIGAMENT CRITERION 'B' MET	Remaining ligament $1.25*[Ka/(YS + YSo)]*2 = 21.5 mm$
Specimen width B = 1.000 inches = 25.4 mm THICKNESS CRITERION *C* MET	Specimen width $[Ka/(YS+YSo)]^2 = 17.2 \text{ mm}$
Crack jump Aa - Ao = 49.2 mm WELD EMBRITTLED SPEC. CRACK JUMP-LEN	ASTM 2N = 22.8 mm NOTH CRITERION "D" MET
Crack jump Aa · Ao = 49.2 mm	Min. crack jump. (Ko/YS)*2/2P1 = 20.7 mm

WELD EMBRITTLED SPEC. CRACK-JUMP LENGTH CRITERION "E" MET

CRACK ARREST COMPUTER CODE CA_TEST Version 5.1 Revised February 14, 1990) Time of this run 12 Sep 1990 16:06:07

SPECIMEN NO. A72W57 (DUPLEX) TEST TEMPERATURE = 21 C from disk CAN02 TEST DATE = 080000

YIELD STRENGTH = 496 MPa

YOUNG'S MODULUS = 206100 MPs INCR. TO YIELD (DYN) (Yso) = 205 MPs

Pre-test specimen dimension measurements

 $\begin{array}{rcrcrcr} B &=& 1.298 \mbox{ inch} \\ BN &=& 976 \mbox{ inch} \\ 2H &=& 6.000 \mbox{ in } \\ WT &=& 6.000 \mbox{ in } \\ WT &=& 6.35 \mbox{ mm} \\ RS &=& 9.40 \mbox{ mm} \\ RT &=& 41.13 \mbox{ mm} \\ TU &=& 27.07 \mbox{ mm} \\ N &=& 1.43 \mbox{ mm} \\ TV &=& 42.66 \mbox{ mm} \end{array}$

Pre-test specimen dimension calculations

Clip gage is located at (FF/W) = .25

EBCL= 58.53 mm

Pre-test loading calculations

1. 1. 1. 1.

This is a DUPLEX specimen, only one loading cycle needed

FOR CYCLE No. = 1, maximum COD = 28.5 mils 2.85 Volts

CRACK ARREST COMPUTER CODE CA_TEST Version 5.1 Revised February 14, 1990) Time of this run: 12 Sep 1990 16:05:03

SPECIMEN NO. A72W57 (DUPLEX) TEST TEMPERATURE = 21 C from disk CAN02 TEST DATE = 000000

Post-test crack arrest measurements

C

a

W-Az AT + BN/4 = 54.341 mm W-Az AT MIDLINE = 54.484 mm W-Az AT -BN/4 = 56.693 mm

Zero load disp. offset at end of cycle 1 $(R1) \approx -7$ mils Total disp. offset at end of cycle n-1 $(R3) \approx 0.0$ mils Displacement at onset of unstable crack growth (P4) = 50.5 mils Displacement approx. 1 s after arrest $(P5) \approx 51.8$ mils

Post-test crack arrest calculations

Length of remaining ligament W-Aa(average) = 55.2 mm

Crack jump Aa - Ao = 29.0 mm

Net COD at initiation per ASTM = 1.283E-03 mNet COD at arrest per ASTM = 1.290E-03 m

* Ko = 216 MPa.m^{0.5} *

* Ka = 146 MPa.m^{*}0.5 *

VALIDITY CRITERIA PER ASTM 1221-88, PARA. 9.3

ACTUAL VALUES	ASTM CRITERIA MINIMUMS				
Length of remaining ligament (W - Aa) = 55.2 mm REMAINING LIGAMENT CRITERION 'A' MET	Remaining ligament (0.15W)	= 19.1 mm			
Length of remaining ligament (W - Aa) = 55.2 mm REMAINING LIGAMENT CRITERION "B" MET	Remaining ligament 1.25*{Ka/(YS	+YSo)]*2 = 54.5 mm			
Specimen width B = 1.298 inches = 33.0 mm **THICKNESS CRITERION 'C' NOT MET***	Specimen width $[Ka/(YS+YSo)]^2$	= 43.6 mm			
Crack jump beyond EB weld (Aa - EBCL) = 13.4 mm ***DUPLEX SPEC. CRACK JUMP CRITERION *D* + *	Specimen width at notch (Bn) = E* NOT MET***	.976 inches = 24.8 mm			

APPENDIX E

STRIP CHARTS AND FRACTURE SURFACES

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Length of Remaining Ligament =

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Specimen Identification	-	A72W10
fest Temperature	-	75°C
Crack Arrest Toughness	-	101 MPa•/m
length of Remaining Ligament	-	49 2 mm





Specimen Identification	-	A72W11
Test Temperature	-	75°C
Crack Arrest Toughness	-	133 MPa•/m
Length of Remaining Ligament	-	61.1 mm



Specimen Identification	-	A72W11		
Test Temperature	-	75°C		
Crack Arrest Toughness	-	133 MPa•/m		
Length of Remaining Ligament	-	61.1 mm		



-12.274


Specimen Identification		A72W13
fest Temperature	-	60°C
Crack Arrest Toughness	-	81 MPa•√m
Length of Remaining Ligament	-	44.2 mm

72w13 ERS 60 70 100 80: 90 to a second s Oak Ridge National Laboratory Menths And Ceramics Division S T

Specimen Identification	-	A72W13
Test Temperature	-	60°C
Crack Arrest Toughness		81 MPa•/m
Length of Remaining Ligamen	t - 1	44.2 mm







100°C 144 MPa+/m 61.5 mm Crack Arrest Toughness Length of Remaining Ligament Specimen Identification Test Temperature





Specimen Identification	10001	E/ZW15
Test Temperature	-	75°C
Crack Arrest Toughness	1	114 MPa-/m
length of Remaining Ligament		47.1 mm

.



Length of Remaining Ligament -

2

47.1 mm











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Length of Remaining Ligament -



Specimen Identification		A/ZW17
Test Temperature		100°C
Crack Arrest Toughness	-	118 MFa•/m
Length of Remaining Ligament	-	53.8 mm





Specimen Identification	-	A72W18
Test Temperature	-	90°C
Crack Arrest Toughness	-	132 MPa*/
Length of Remaining Ligamen	t -	57.0 mm



A72W18 90°C 132 MPa∗√m 57.0 mm











MILLIMETERS 10 30 40 60 70 80 30 100 manthemalounder and an and Gas Riciga Mananal Laboratory Matals And Caramics Division 72W23 A72W23 60°C

Specimen identification		A/ 6W63
Test Temperature	-	60°C
Crack Arrest Toughness		74 MPa•√m
Length of Remaining Ligament		22.0 mm





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TERS 19 20 30 40 50 60 70 80 50 10 100 Serve Printight Research all Latters artigry Metals And Genemics Dics on 72W27 Specimen Identification A72W27 Test Temperature 29°C

61 MPa ./m

18.3 mm

Crack Arrest Toughness

Length of Remaining Ligament -









Specimen Identification		A72W30
Test Temperature		+25°C
Crack Arrest Toughness	-	43 MPa∗√m
Length of Remaining Ligament		15.4 mm









Specimen Identification	100	A73W13
Test Temperature	-	75°C
Crack Arrest Toughness	-	150 MPa•/m
Length of Remaining Ligament		65.8 mm



Specimen Identification = Test Temperature = Crack Arrest Toughness = Length of Remaining Ligament =

A73W13 75°C 150 MPa•/m 65.8 mm




Specimen Identification	-	A73W14
Test Temperature	-	90°C
Crack Arrest Toughness	-	159 MPa ./
Length of Remaining Ligament	-	29.1 mm



Specimen Identification		A73W14
Test Temperature	-	90°C
Crack Arrest Toughness	-	159 MPa•/
Length of Remaining Ligament	-	29.1 mm





45.6 mm

Length of Remaining Ligament =



SPEC # A DATE: DAT	Normal/Inverted Non -Irradiated	
TEST TEMP. Contraction VOLTAGES: # CAGØL	MACHINE SETTINGS Load Range = ± 20 × 17	
Excitation - 6.535	Strain Range = 1.249 W	
During calibration-	Stroke Range = + /.	
0 0.07 75 - 7.55 25 -2.52 100 - 10.0* 50 -5.52	X-Y CHART SETTINGS X= 1 1/2	
In specimen at zero- 200 (a. 1)	¥= / 1/	

13

.2.52 V



-5.13 4





Specimen Identification	-	A73W17 (no test)
Test Temperature	-	100°C
Crack Arrest Toughness	-	
Length of Remaining Ligamen	t	-



SPEC. # A 73w 17 DATE: 10-25-89 Normal (Inverted Non -Irradiated) TEST TEMP. 30°C CLIP GAGE # CA6 # CA6 # C MACHINE SETTINGS Load Range = 1 20 FIPS/50KIP. VOLTAGES: Strain Range = = 2,0% Excitation - 6 588 Stroke Range = ±1 During calibration-75 -7.55 0 0.0 X-Y CHART SETTINGS 25 -2.52 100 -12.10 CLANGED TO SO KIP X= 1 % 50 -5.07 Y= 1 1/2 2/ 1/11 In specimen at zero- -3.77 (0.0) 12,000 /24,000 1000/20,000 1/2 5000/15,000 der 600/2,000 440/8,000 シーシー 2-----0 0 MILS

135



Specimen Identification	-	A73W18
Test Temperature	-	75°C
Crack Arrest Toughness	-	131 MPa•√m
Length of Remaining Ligament	-	51.3 mm



Specimen Identification	-	A73W18
fest Temperature	-	75°C
Crack Arrest Toughness	-	131 MPa•/m
Length of Remaining Ligament	÷.	51.3 mm





Oak Ridge National Laboratory

Meths And Construct Division

Specimen Identification	-	A73W21
Test Temperature	-	75°C
Crack Arrest Toughness	-	107 MPa•/m
Length of Remaining Ligament	-	54 1 mm



Oak Ridge National Laboratory

A73W21 Specimen Identification

75°C Test Temperature 107 MPa ./m Crack Arrest Toughness Length of Remaining Ligament -54.1 mm





Test Temperature = Crack Arrest Toughness = Length of Remaining Ligament =

90°C 180 MPa•√m 45.3 mm













57 MPa•√m 21.5 mm

fest Temperature	
Crack Arrest Toughness	-
Length of Remaining Ligament	

145 (V. F. Brief Contraction Contraction Con R 4 No. Se Calmerer . 40 36 30 - 700 20,2 X- 5 % A TENS 29°C. 23 MILES - 1950 Jac A SPIC 18.18 ... 20 Marind Jan Pr. Jully Cope 2 436 2 20 * 4 * à 5 #20890 #10 K 3(#14) 2 SPEC. A 73 ,131 244E 8-23-75 5 CAN IN KANCE LOND RANGE Dares Chicker ?! 0 04 14 5 12 0



ameronant

















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Specimen Identification	-	A73W45
Test Temperature	-	90°C
Crack Arrest Toughness	-	114 MPa*/m
Length of Remaining Ligament	-	55.3 mm









Length of Remaining Ligament -

47.1 mm





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The objective of this irradiation on the shift and data. Two submerged-arc we were commercially fabricate fabricated from these welds 288°C to an average fluence Evaluation of the results is arrest toughness temperatur impact temperature shift a bound curves (for the rang- have been altered by irrad	study was to determine the effect nd shape of the lower-bound curve elds with copper contents of 0.23 ed in 220-mm-thick plate. Crack- s were irradiated at a nominal te e of 1.9×10^{19} neutrons/cm ² (>1 M shows that the neutron-irradiation re shift is about the same as the t the 41-J energy level. The shape of test temperatures covered) of iation compared to those of the A	t of neutron to crack-arrest and 0.31 wt % arrest specimens imperature of deV). on-induced crack- s Charpy V-notch ape of the lower- did not seem to ASME K _{I*} curve.
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