



SACRAMENTO MUNICIPAL UTILITY DISTRICT ☐ 6201 S Street, Box 15830, Sacramento, California 95813; (916) 452-3211

February 3, 1982

DIRECTOR OF NUCLEAR REACTOR REGULATION
ATTENTION JOHN F STOLZ CHIEF
OPERATING REACTORS BRANCH 4
U S NUCLEAR REGULATORY COMMISSION
WASHINGTON DC 20555



DOCKET 50-312
RANCHO SECO NUCLEAR GENERATING
STATION UNIT NO 1
HIGH PRESSURE INJECTION NOZZLE
CYCLES

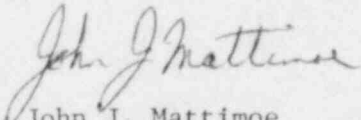
On December 11, 1981, the District responded to the Atomic Safety and Licensing Appeal Board's Memorandum and Order, ALAB-655. Since that time, we have had numerous discussions with your staff concerning the allowable number of transient cycles for the high pressure injection nozzles. We have provided the Babcock & Wilcox Company Field Change Package, "HPI Nozzle", Document No. 04-3370-00 FC-0174-00, for your review, and following additional questioning, B&W Calculation Nos. 32-1121811-00, "HPI Nozzle Usage Factor" and 32-1119809-01, "HPI Nozzle Usage Factor". These documents form the basis for including within the total number of allowable reactor trips (400) a specific provision for up to 70 actuations of high pressure injection following such a trip. (These 70 cycles are in addition to the 80 cycles included in the original stress analysis.) As noted on page 4 of Document 32-1121811-00, the original 80 cycles resulted in a usage factor of only .44, which demonstrates the conservatism in the analysis. These documents are attached to this letter for your reference.

As discussed in our response to ALAB-655, these analyses do not explicitly address the Appeal Board's question concerning the ultimate number of cycles these nozzles are physically capable of safely withstanding. Our response provided the justification for not responding to this question. We feel, however, that these additional cycles demonstrate that safety limits are not being approached, and that no safety concern exists due to cycle usage at Rancho Seco Unit No. 1. Your staff has requested additional information for their use in reviewing these calculations. The workload within the Babcock & Wilcox Company (due to such factors as the Thermal Shock issue) precludes this information being generated and submitted for your review prior to July 1, 1982.

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PDR ADOCK 05000312
P PDR

A001
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1/1

This information relates primarily to techniques used in the original Rancho Seco Stress Analysis and we feel this information is irrelevant to the Appeal Board's concern over high pressure injection nozzle usage, but should satisfy your needs in reviewing this change to the Rancho Seco Administrative Procedures. We expect to receive a written request for information from you which will confirm the information to be supplied in July of this year.



John J. Mattimoe
Assistant General Manager
and Chief Engineer

Attachments

cc: Mr. Tom Baxter - Shaw, Pittman, Potts & Trowbridge
Mr. Dave Holt - Babcock & Wilcox Company
Mr. Mark Padovan - Nuclear Regulatory Commission

T-143

DOCUMENT SUBMITTAL FORM

0 SMUD
 6201 S Street
 Sacramento, CA 95813
 TTN Mr. R.J. Rodriguez

DATE 10/22/80
 B&W CONTRACT NO. 620-0011
 CUST. SMUD
 CUST. ORDER NO. GRS95243
 SHEET 1 OF 1
 TYPE DOC Field Change Package

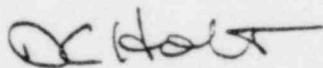
2 pkgs. ☐ UNDER SEP COVER ☒ ENCLOSED
☐ UNDER SEP COVER ☐ ENCLOSED

REFERENCES ☐ FOR COMMENTS & APPROVAL BY
☐ FOR INFORMATION ONLY
☐ HAS BEEN REVISED AS PER YOUR PREVIOUS COMMENTS OF
☒ FOR FINAL DISTRIBUTION
☐ FURTHER EXPLANATION REQUIRED
 SEE BELOW OR ATTACHMENTS

DOCUMENT DESCRIPTION

GIN	SUPPLIER/CUSTOMER DOC. NO.	B&W DOC. NO.	PART/ TASK NO.	GRP. NO.	SEQ. NO.	DOC. TITLE
&W		04-3370-00 FC-0174-00	50	01		HPI Nozzle

c: R.P. Oubre
 D.G. Raasch
 R.A. Dieterich
 J.V. McColligan
 L.G. Sweiger
 J.T. Janis



D.C. Holt
 Engineering Product Manager

FIELD CHANGE AUTHORIZATION

FC-0174-00

BABCOCK & WILCOX

CUSTOMER SMUD	ORIGINATOR K.C. Smith	DATE 6-11-80	CONT. NO. 620-0011	DOC. I.D. NO. 04-3370-00	REV. NO.
SUPPLIER Babcock & Wilcox	P.A. NO.		PART NO., OR. TASK NO.	GROUP NO.	SEQ. NO.
TITLE (MAX 30 CHARACTERS) HPI Nozzle			50	001	001

DESCRIPTION AND JUSTIFICATION OF CHANGE:

It has been reported that the SMUD operators have been starting a second make-up pump following each reactor trip to obtain additional make-up flow. This was done to prevent loss of indicated pressurizer level during the transient. All four of the high pressure injection nozzles have been used for these occurrences and the three nozzles which are not continuously adding make-up flow receive a thermal shock from the cold BWST water.

The contract Functional Specification, CS(F)3-92/NSS-11/0372, defines forty cycles of high pressure injection actuation and forty cycles of high pressure injection testing. At SMUD's present usage rate, 31 cycles, the nozzles would exceed their allowable cycles before the end of their forty-year life.

B&W has performed a HPI nozzle stress analysis evaluation of the actual experienced transients (actual transients are less severe than functional specification transients) to determine their actual fatigue life expectancy. In reviewing the SMUD HPI Design Analysis, it was discovered that the original analysis was performed as a generic analysis and the fatigue analysis was performed to a later, more restrictive design code. In analyzing the HPI nozzle to the contractual design code, 1968 draft USAS, B31.7 with June 1968 errata, the number of allowable HPI actuations can be approximately doubled. The re-analysis must conform to the requirements of ASME Section XI which now applies to the site and allows the use of earlier codes.

This Field Change Authorization is being written to document the increase in allowable HPI system actuation cycles and therefore no hardware changes are required.

REASON FOR CHANGE:

- ☒ SITE PROBLEM
☐ ENGINEERING REQUIREMENT
☒ CUSTOMER REQUEST
☐ IMPROVEMENT
☐ OTHER (SPECIFY) _____

TASKS AFFECTED	NO.	COMPLETED	DATE
TASK NO.	TASK TITLE	DOCS	(T.E. SIGN)
17	RCS Comp.	0	K.C. Smith 10/10/80
50A	RC Pipe-Sys. Des.1	1	K.C. Smith 10/10/80
50	RC Piping	1	P.B. Reynolds 10/10/80
10	Proj. Engr.	0	C.H. Hudson 10/13/80
09	Power Systems & Controls	1	H.B. Ratz 10/14/80

SPR.DOC.. ~~XXXXXXXXXXXX~~ 13-11-361-00

CONCUR TO PROCEED	APPROVALS NAME	DATE	CUSTOMER/CUSTOMER AGENT DISPOSITION OF FIELD CHANGE:
KCS	K.C. Smith	10/10/80	☐ IMPLEMENTED ☐ NOT IMPLEMENTED
RRR	P.B. Reynolds	10/10/80	REF. _____
CTH	C.H. Hudson	10/13/80	PROJECT MANAGER _____ DATE _____
CH	H.B. Ratz	10/14/80	AUTH. CHANGE NO. _____ PCA NO. _____
			SHEET 1 OF 10

FIELD CHANGE AUTHORIZATION - AFFECTED DOCUMENTATION

[illegible]

FIELD CHANGE AUTHORIZATION-CONTINUATION

☐ EXPEDITE
☒ NORMAL

REFERENCE DOCUMENT TITLE

REF DOCUMENT NO.
CS(F)3-37/NSS-11/0372

Specification for Reactor Coolant System Piping

SHEET 1 OF 1

DESCRIPTION OF CHANGE:

1. Revise paragraph 1.1 of Appendix 1 of the reference document to read as follows:

1.1 General Functional Specification for Reactor Coolant System Components, Specification No. CS(F)-3-92/NSS-11/0672 as modified by Field Change Authorization 04-3370-00.

2. Add paragraph 2.2 in Appendix 1 of the reference document to read as follows:

2.2 ASME Boiler and Pressure Vessel Code

a. Section XI, Inservice Inspection

1974 Edition
with Summer
1975 Addenda

The reanalysis of the HPI nozzle is to be in accordance with the requirements of the code listed above.

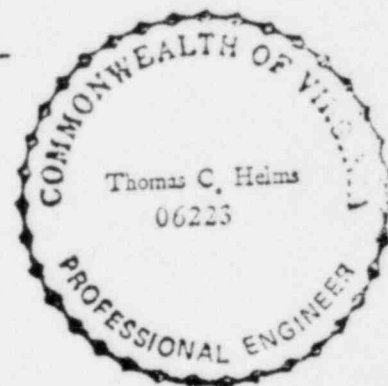
The changes to the reference document above are certified to be correct and complete and in compliance with the requirements of ASME Boiler and Pressure Vessel Code, Section XI.

Thomas C. Helms PROFESSIONAL ENGINEER

Virginia
STATE

06223
REGISTRATION NO.

10/10/80
DATE



REVISED REQUIREMENTS (CONTINUATION SHEET)	SUPPLIER REVISIONS	TASK NO.	TASK TITLE
PREPARED BY	REVIEWED BY	50A	R.C. Pipe - System Designer
DATE	DATE	FCA NO.	
REVIEWED BY	DATE	04-3370-00	
APPROVED BY	DATE	SHEET	3 OF 10

FIELD CHANGE AUTHORIZATION-CONTINUATION

☐ EXPEDITE
☒ NORMAL

REFERENCE DOCUMENT TITLE

GENERAL FUNCTIONAL SPECIFICATION
FOR REACTOR COOLANT SYSTEM

REF DOCUMENT NO.

CS(F) 3-92/NES-11/672

SHEET 1 OF 6

DESCRIPTION OF CHANGE:

Change the RCS Functional Specification as per FCA continuation sheets 2 through 6.

The changes to be incorporated in this revision (03) of the spec include:

1. A revision to Section 6.9, Transient 8- Reactor Trip. Specifically, paragraphs will be added to sections 6.9.2 and 6.9.3.1 of the text to provide for manual actuation of the high pressure injection system after a reactor trip.
2. Table 2 will be revised to include 70 design cycles for the above noted change.
3. Figure 8-14 will be added to the spec.

REVISED REQUIREMENTS (CONTINUATION SHEET)		SUPPLIER REVISIONS		TASK NO.	TASK TITLE
PREPARED BY	DATE	REVIEWED BY	DATE	09	POWER SYSTEMS + CONTROLS
R. B. Park	8/15/80			FCA NO.	
REVIEWED BY	DATE			04-	3370-00
APPROVED BY	DATE			SHEET	4 OF 6
J. H. Cullen	8/15/80				

Figure No.Title

8-8	Surge Line Temperature and Flow.
8-9	Makeup and Spray Flow Rate and Temperature.
8-10	Reactor Trip, Cooldown and Normal Power Recovery - Primary System.
8-11	Reactor Trip, Cooldown and Normal Recovery - Secondary System.
8-12	Reactor Trip from Full Power for Trans. 8A&B - Steam Gen. Conditions.
2 8-12A	Feedwater Conditions Following Reactor Trip (8A&B).
8-13	Loss of Feedwater Reactor Trip-Steam Feedwater System Parameters.
ABP 8/15/80 8-14	<u>MAKEUP FLOW AND TEMPERATURE - MANUAL ACTUATION OF HIGH PRESSURE INJECTION SYSTEM</u> <u>Transient No. 9 - Rapid Depressurization</u>
9-1	Reactor Coolant Conditions.
9-2	Surge Flow and Temperature.
9-3	Makeup and Spray Flow and Temperature.
9-4	Feedwater and Reactor Coolant Flow Range.
9-5	Steam Temperature and Test.
9-6	Steam Generator Pressure.
9-7	Feedwater Temperature Range.
	<u>Transient No. 10 - Change of Flow</u>
10-1	Reactor Coolant Temperature.
10-2	Surge Line Water Temperature and Flow Rate.
10-3	Makeup and Spray Line Flow and Water Temperature.
10-4	FW-Steam System Parameters for Steam Generator with Two Reactor Coolant Pumps.
10-5	FW-Steam System Parameters for Steam Generator with One Pump.
	<u>Transient No. 11 - Rod Withdrawal Accident</u>
11-1	Reactor Coolant Temperature and Pressurizer Liquid Level and Pressure.
11-2	Spray and Makeup Temperature and Flow Rate.

2

in Transient 3 (Plant loading 8% to 100%). These return to power events are included in the 18000 cycles specified for Transient 3.

*Insert indicated
addition shown on
following page.
RBP
8/15/80*

- 2) 56 Reactor trip events, followed by cooldown as described in Transient 1B (Cooldown), followed by heatup as described in Transient 1A (heatup) and Transient 3 (Plant Loading). These 56 heatup and cooldown cycles following reactor trip are included in the 240 cycles specified for Transient 1.

6.9.3 Transient Data

6.9.3.1 Reactor Coolant System

Reactor Coolant System conditions for the reactor trip transients are presented in the following figures:

Type A Reactor Trip - Figures 8-1, 8-1A, 8-2, 8-3
Type B Reactor Trip - Figures 8-4, 8-4A, 8-5, 8-6, 8-10
Type C Reactor Trip - Figures 8-7, 8-7A, 8-8, 8-9

*Insert indicated
addition shown on
following page.
RBP
8/15/80*

6.9.3.2 Steam Generators

Steam Generator conditions for the reactor trip transients are presented in Figures 8-11, 8-12, 8-12A and 8-13.

2 | 6.10 Transient 9 - Rapid Depressurization (Upset Condition)

6.10.1 General Description

Rapid depressurization is a short term, rapid cooling of the Reactor Coolant System by the steam generators in order to reduce the reactor coolant system pressure to a value less than the design pressure (1065 psia) of the steam generators within 15 minutes. The objective of the rapid depressurization is to isolate a tube leak.

The initial conditions at the start of the transient are assumed to be hot standby with decay heat removal by the steam generators dumping steam to the condenser. The turbine bypass control pressure is assumed to be 1050 psia. This gives an average reactor coolant system temperature of about 558.7°F.

When average reactor coolant temperature has been reduced to 500°F, and reactor coolant pressure is equal to or less than 1065 psia (15 minutes), normal cooldown as described in Transient 1B is initiated.

6.10.2 Cycles

The number of rapid depressurization events for design shall be 40. A complete cycle consists of a) power reduction from 100% power as described in Transient 4 (Plant Unloading); b) Rapid de-

1

Insert in §6.9.2
RBP
8/15/80

Manual Actuation of High Pressure Injection System After Reactor Trip

Occasionally after a reactor trip, it may be necessary to manually actuate the high pressure injection system in order to obtain additional makeup flow. This should only be done to prevent the loss of indicated pressurizer level during the transient. For the total of 400 reactor trip events, an allowance of 70 cycles of high pressure injection system actuations shall be considered for design purposes.

Insert in §6.9.3.1
RBP
8/15/80

Manual Actuation of High Pressure Injection System After Reactor Trip - Figure 8-14

Table 2Operating Transient Cycles

<u>Transient Number</u>	<u>Transient Description</u> (ASME Category)	<u>Design Cycles</u>	<u>Figure Nos.</u>
2	1A Heatup from 70°F to 8% Full Power (Normal)	240	1A-1 - 1A-1
	1B Cooldown from 8% Full Power (Normal)	240	1B-1 - 1B-1
2	Power change 0 to 15% and 15 to 0% (Normal)	1440	2A-1 - 2A-5 2B-1 - 2B-5
2	3 Power Loading 8% to 100% power (Normal)	18,000	3-1 - 3-5
	4 Power Unloading 100% to 8% power (Normal)	18,000	4-1 - 4-5
5	10% Step Load Increase (Normal)	8,000	5-1 - 5-4
6	10% Step Load Decrease (Normal)	8,000	6-1 - 6-5
7	Step Load Reduction (100% to 8% Power) (Upset)		
	Resulting from turbine trip	160	7-1 - 7-5
	Resulting from electrical load rejection	150	
	Total	310	
8	Reactor Trip (Upset)		
	Type A	40	8-1 - 8-13
	Type B	160	
	Type C	88	
	Trips included in transient numbers 11, 15, 16, 17 & 21	112	
	Total *	400	
9	* MANUAL ACTUATION OF HIGH PRESSURE INJECTION SYSTEM AFTER REACTOR TRIP (Upset)	70	8-14
10	Change of Flow (Upset)	20	10-1 - 10-5
11	Rod Withdrawal Accident (Upset)	40	11-1 - 11-4
2	12 Hydrotests (Test)	35	
13	Steady-State Power Variations (Normal)		13-1
14	Control Rod Drop (Upset)	40	14-1 - 14-4
15	Loss of Station Power (Upset)	40	15-1 - 15-7
16	Steam Line Failure (Faulted)	1	16-1 - 16-3
17A	Loss of Feedwater to One Steam Generator (Upset)	20	17A-1 - 17A-1
17B	Stuck Open Turbine Bypass Valve (Emergency)	10	17B-1 - 17B-1

RBP
8/15/80

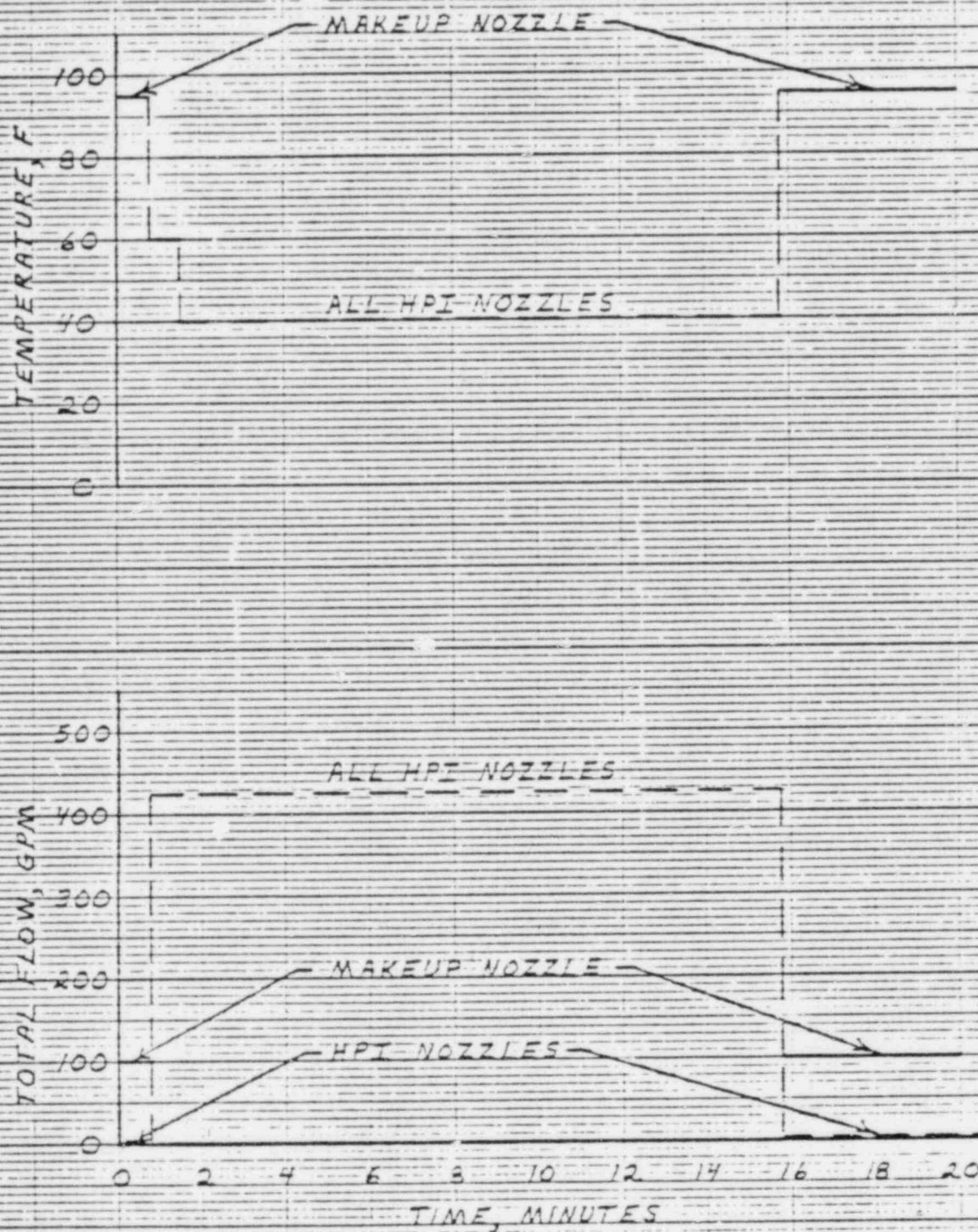


FIGURE 8-14

TRANSIENT NO. 8 (REACTOR TRIP)
 MAKEUP FLOW AND TEMPERATURE - MANUAL
 ACTUATION OF HIGH PRESSURE INJECTION SYSTEM

FIELD CHANGE AUTHORIZATION-CONTINUATION

☐ EXPEDITE
☒ NORMAL

REFERENCE DOCUMENT TITLE

REF DOCUMENT NO.
NONE

STRESS REPORT FOR PRIMARY PIPING (620-0011-50)

SHEET 1 OF 1

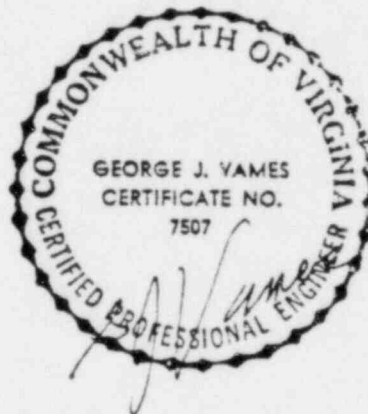
DESCRIPTION OF CHANGE:

The following documents are added as addenda to the original stress report:

32-1119809-01 HPI Nozzle Usage Factor

32-1121811-00 HPI Nozzle Usage Factor

I certify that the two (2) above mentioned analysis when taken as a whole justify the modifications to be made under the authority of FCA No. 04-3370-00 and that these two (2) analysis when considered as a whole meet the requirements imposed by the applicable Equipment Specification, as modified by FCA No. 04-3370-00. Attested to this date: October 10, 1980



George J. Vames
 Babcock & Wilcox
 Nuclear Power Generation Division
 Lynchburg, Virginia

License No. 7507
 Virginia State Board
 of Professional
 Engineers

REVISED REQUIREMENTS (CONTINUATION SHEET)		SUPPLIER REVISIONS		TASK NO.	TASK TITLE
PREPARED BY	DATE	REVIEWED BY	DATE	50	Reactor Coolant Piping
RR Schaefer	10/9/80	RB Reynolds	10/10/80	FCA NO.	
REVIEWED BY	DATE			04- 3370-00	
G.J. Vames	10-10-80	W.H. Sims	10/10/80	SHEET 1 OF 10	
APPROVED BY	DATE				
G. R. Baker	10-10-80				

CONTRACT/STANDARD NO.

620-0011

DOCUMENT RELEASE

NOTICE (DRN)

RELEASE DATE PAGE

OCT 13 1980 1 of 1

PART-MARK/TASK-
GROUP-SEQ.

B&W DOCUMENT NO.

DOCUMENT TITLE

FUL
STAT.

REL. NO.

50/001/001 32-1121811-00 HPI NOZZLE USAGE FACTOR ~ ~

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R.R. SCHAEFER

1 1

K.C. SMITH

1 1

ORIGINAL

R.R. Schaefer 10/9/80
NAME DATE

REVIEWED BY

A.D. McKim 10/9/80
NAME DATE

CALCULATION DATA/TRANSMITTAL SHEET

DOCUMENT IDENTIFIER

TYPE: RESEARCH & DEVELOPMENT SAFETY ANALYSIS REPORT NUC. SERV. INPUT LESSON LGT. X DISCERN VERIF.
OTHER

TITLE HPI Nozzle Usage Factor

PREPARED BY R.R. Schaefer REVIEWED BY Alvin D. McKen
TITLE Senior Engineer DATE 10/9/80 TITLE Supervisory Engineer DATE 10/9/80

PURPOSE:

To determine if the HPI nozzle can withstand 70 HPI actuations following a reactor trip condition. These will be in addition to the 40 rapid depressurization and 40 test transient cycles specified in the contract functional specification CS(F)3-92/NSS-11/0372.

SUMMARY OF RESULTS (INCLUDE DOC. ID'S OF PREVIOUS TRANSMITTALS & SOURCE CALCULATIONAL PACKAGES FOR THIS TRANSMITTAL)

The HPI Nozzle can withstand 40 test transient cycles, 40 rapid depressurization transients and 70 reactor trip transients as described in FCA 04-3370-00.

DISTRIBUTION

BABCOCK & WILCOX
GENERAL CALCULATIONS

32
DOC. I.D.

1121811
SERIAL NUMBER

00
REV

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OWNER	PROP NO	FILE NO
PROJECT	DWG NO	COMP NO
UNIT NO 70.00	GROUP NO	
DESIGNED BY RPS	DATE 8/27/80	SHEET NO 20/18
CHECKED BY LBM	DATE 10/5/80	

GENERAL CALCULATIONS

PURPOSE

The purpose of this calculation is to amend CDS 32-1119809-00, Ref (1). Ref (1) justified an additional 70 cycles of rapid depressurization transient. This calculation will justify 70 cycles of HPI actuation following a reactor trip instead of 70 additional rapid depressurization transient cycles was done in Ref (1).

DISCUSSION

CDS 32-1119809-00, Ref. (1), determined the maximum allowable number of rapid depressurization transient cycles. This calculation was based upon the assumption that FCA 04-3370-00 would characterize the 70 additional HPI transients as required by SPR #13-11-361-00, Ref. (5) as rapid depressurization transients.

When FCA 04-3370-00 was issued the additional 70 HPI transients were assumed to follow a reactor trip transient. Thus, this calculation will show that the HPI can withstand 70 cycles of HPI Actuation following a reactor trip transient instead of 70 additional rapid depressurization cycles as shown in Ref. (1).

POWER <i>SMUD</i>	PROP NO	CONT NO <i>620-0011</i>
	DWG NO	FILE NO
	COMP NO <i>50</i>	GROUP <i>0</i>
BY <i>RRS</i> DATE <i>10/9/80</i> CHD BY <i>ATM</i> DATE <i>12/5/80</i>		SHEET NO <i>3 of 18</i>

BABCOCK & WILCOX
GENERAL CALCULATIONS12
DOC ID.1121811
SERIAL NUMBER00
REVRESULTSMAXIMUM PRIMARY + SECONDARY STRESS
INTENSITY RANGE

$$S = 123.9 \text{ KSI} > 35_{\text{M}} \text{ (PG. 5, REF [1])}$$

THEREFORE AN ELASTIC-PLASTIC
FATIGUE ANALYSIS WAS PERFORMED.USAGE FACTOR = 0.441 < 1.0 FOR 40
TEST TRANSIENTS AND 40 RAPID
DEPRESSURIZATION TRANSIENTS.
(PG. 5, REF. [1])USAGE FACTOR = 0.26 FOR 70 NPI
ACTUATIONS FOLLOWING A REACTOR
TRIP TYPE TRANSIENT.

TOTAL CUMMULATIVE USAGE FACTOR IS;

$$U_T = 0.44 + 0.26 = 0.70 < 1.0$$

(PG. 16)

CONCLUSIONTHE NPI NOZZLE CAN WITHSTAND
40 TEST TRANSIENT AND 40 RAPID
DEPRESSURIZATION TRANSIENT CYCLES.
IN ADDITION THE NPI NOZZLE
CAN WITHSTAND 70 ADDITIONAL
CYCLES OF NPI ACTUATION FOLLOWING
A REACTOR TRIP TRANSIENT AS
DESCRIBED IN REF. [7].

CUST. NO.

PROP. NO.

FILE NO.

SUF.

DWG. NO.

COMP. NO.

CONT. NO. 620.00

GROUP NO.

CALC. BY

RRS

DATE 10/9/93

SHEET NO. 4 of 15

ADAL

10/9/93

BABCOCK & WILCOX
GENERAL CALCULATIONS32
DOC ID1121811
SERIAL NUMBER00
REVTHERMAL PARAMETERS

PAGE 4-2 OF REF [2] GIVES THE FILM COEFFICIENT VALUES USED IN THE ANALYSIS FOR THE RAPID DEPRESSURIZATION TRANSIENT. THIS FILM COEFFICIENT CALC. ASSUMED A FLOW RATE OF 425 GPM THROUGH EACH NOZZLE. THE ACTUAL FLOW IS 425 GPM TOTAL FLOW SPLIT EQUALLY BETWEEN ALL 4 HPI NOZZLES. THUS, THE FILM COEFFICIENT SHOULD HAVE BEEN;

IN BRANCH (PER PG. 4-2, REF. [2])

FOR 60°F WATER

$$C_p = 1, \quad \rho = 62.34 \text{ LB/FT}^3, \quad \mu = 2.71$$

$$K = 0.344, \quad D = 0.177 \text{ FT}, \quad A = 0.0246 \text{ FT}^2$$

$$hD^{0.2} = 0.023K \left(\frac{G}{\mu} \right)^{0.8} \left(\frac{C_p \mu}{K} \right)^{0.4}$$

$$G = \left(425 \text{ GPM} / 4 \right) \left(60 \frac{\text{MIN}}{\text{HR}} \right) \left(0.1337 \frac{\text{FT}^3}{\text{GAL}} \right) \left(62.34 \frac{\text{LB}}{\text{FT}^3} \right)$$

$$0.0246 \text{ FT}^2$$

$$G = 2159948 \text{ LB/FT}^2\text{-HR}$$

$$hD^{0.2} = 0.023 \left[\frac{2159948}{2.71} \right]^{0.8} \left[\frac{(1.0)(2.71)}{0.344} \right]^{0.4} (0.344)$$

$$hD^{0.2} = 950.8$$

$$h = 1345 \text{ BTU/HR-FT}^2\text{-°F}$$

$$h = 4100 \text{ BTU/HR-FT}^2\text{-°F (USED IN REF. [1] & [2])}$$

CUSTOMER

PROJ. NO.

FILE NO.

SL

DWG. NO.

COMP. NO.

CONT. NO. 620-00

GROUP NO.

CALC. BY RRS

DATE 10/7/80

SHEET NO. 5 of 18

RABCOCK & WILCOX
 GENERAL CALCULATIONS

 32
 DOC ID

 1121811
 SERIAL NUMBER

 00
 REV.

THE FILM COEFFICIENT USED IN REFERENCES [1 & 2] FOR THE TEST TRANSIENT IS 1300 BTU/HR-FT²-°F. BOTH OF THESE TRANSIENTS INITIATE AT 550 °F.

FROM PAGE 4-11, REF. [2], ITTERS 3, 6 & 8 EACH OCCUR IN THE FIRST TEN SECONDS. FROM PAGE 6 OF REF. [1] THE MAXIMUM STRESS INTENSITY OCCURS AT ITERATION 8 WHICH HAS A TIME POINT EQUAL TO 5.7 SECONDS. THUS, THE MAXIMUM STRESS INTENSITY AT THE NOZZLE END FOR THE TEST TRANSIENT IS FULLY DEVELOPED BEFORE THE TRANSIENT ENDS.

THUS, THE STRESS INTENSITY CALCULATED FOR THE TEST TRANSIENT INITIATION IS ESSENTIALLY WHAT THE ACTUAL RAPID DEPRESSURIZATION TRANSIENT SHOULD HAVE BEEN. THE ONLY SIGNIFICANT DIFFERENCE BETWEEN THE TWO TRANSIENTS IS THE RETURN TO POWER AND TEMPERATURE. THIS PORTION OF THE ANALYZED RAPID DEPRESSURIZATION TRANSIENT IS CORRECT.

HOWEVER, FOR THE PURPOSES OF THIS CALCULATION, THE TEST TRANSIENT CYCLES (40 CYCLES) AND THE RAPID DEPRESSURIZATION TRANSIENT CYCLES (40) WILL BE JUSTIFIED BY LDS-32-1119809-00. THEY HAVE A CUMULATIVE USAGE FACTOR OF $U=0.441$, AS SHOWN ON PAGE 23 OF REF. [1].

CUSTOMER	PROP NO	FILE NO
SUBJECT	DWG NO	COMP NO
QNT NO 820.00	GROUP NO	
ALC BY RRS	DATE 10/8/20	SHEET NO 6 of 13

BARCOCK & WILCOX
GENERAL CALCULATIONS32
DOC ID1121811
SERIAL NUMBER00
REV

THE ADDITIONAL 70 CYCLES OF HPI
ACTIVATION FOLLOWING A REACTOR TRIP
WILL BE JUSTIFIED BY ADJUSTING
THE STRESS INTENSITY RANGES DISCUSSED
FOR THE TEST AND RAPID DEPRESSURIZATION
TRANSIENTS PREVIOUSLY. CALCULATIONS
AND DATA ARE PRESENTED ON THE
FOLLOWING PAGES FOR EACH TRANSIENT
JUSTIFICATION.

CUSTOMER

SUPERVISOR

LGNT NO 820 00

CALC BY RRS

DATE 1/10/88

PROP NO

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BABCOCK & WILCOX
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TABLE OF REACTOR TRIP CONDITIONS
(PER REF. 3)

TRANSIENT	HIGHEST TEMP	INITIAL TEMP. AT 45 SEC	RETURN TEMP	PRESSURE AT 45 SEC	RETURN PRESSURE
8A	570	550	550	2200	2150
8B	575	550	550	1850	2200
8C	590	575	565	2200	2200
11	592	585	540	2300	1800
15	575	550	550	2200	2150
16	575	560	560		
17A	577	565	550	2100	1850
17B	565	565	560	2450	2200
21	575	575	300	1600	0

FROM THE ABOVE TABLE, TRANSIENT 11 EXHIBITS THE HIGHEST INITIAL TEMPERATURE AND PRESSURE. THE HIGHEST RETURN TEMPERATURE IS 565. THE JUSTIFICATION WILL CONSERVATIVELY USE THE MAXIMUM VALUES.

CUSTOMER

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BARCOCK & WILCOX
GENERAL CALCULATIONS

12

DOC 10

SERIAL NUMBER

REV

TO JUSTIFY THE ANALYSIS OF THE 20 CYCLES FOR THE NUCLEAR STARTING TEMPERATURE, THREE ANALOGOUS THERMAL STRESS RUNS WERE MADE. BAY COMPUTER PROGRAM F91232 WAS USED. USING THIS SLAB TEMPERATURE PROGRAM, THE ARE RAPID DEPRESSURIZATION AND TEST TRANSIENTS WERE ANALYZED AS TO THEIR EFFECT ON THE NOZZLE END THERMAL STRESS WITHOUT DISCONTINUITY EFFECTS. THE RESULTS OF THIS ANALYSIS ARE CONTAINED ON REF FINE SHEETS AND ARE TABULATED BELOW.

CASE	INITIAL	ΔT	FILM	COEFF.	ΔT RATIO	LINEAR	THERMAL	STRESS	STRESS RATIO
1	550	490	4100	1.00	73106	1.00			
2	570	510	4100	1.04	76023	1.04			
3	585	525	4100	1.07	79204	1.07			
4	550	490	1300	1.01	57131	1.00			
5	570	510	1300	1.04	59350	1.04			
6	585	525	1300	1.07	61005	1.07			

NOTES (1) ΔT RATIO = ΔT/ΔT₁ = ΔT/490

(2) STRESS RATIO = S_{20P3}/S₁

OR = S_{50P6}/S₄

CUSTOMER

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BABCOCK & WILCOX GENERAL CALCULATIONS

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 DOW 10

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REV

TO JUSTIFY THE ANALYSIS OF THE 70 CYCLES FOR A HIGHER RETURN TEMPERATURE, THE FOLLOWING RESULTS FROM P91232 ARE TABULATED. THESE RESULTS ARE FROM REF. FIGURE SNAF06E.

CASE	RETURN TEMP	ΔT (T-40)	FILM COEFF.	$\Delta T^{(1)}$ RATIO	LINEAR THERMAL STRESS	STRESS ⁽²⁾ RATIO
1	550	510	35	1.00	-5653	1.00
2	570	530	35	1.04	-5876	1.04
3	585	545	35	1.07	-6039	1.07

FROM THE TABULATIONS ON PAGE 6 AND ABOVE, THERE IS SUFFICIENT JUSTIFICATION FOR USING A ΔT RATIO METHOD TO JUSTIFY THE DIFFERING TEMPERATURE FROM SPP #13-11-361-00 VERSUS THE RAPID DEPRESSURIZATION TRANSIENT ANALYZED. IN ORDER TO CALCULATE A STRESS INTENSITY RANGE FOR THE REACTOR TRIP FOLLOWED BY MPI ACTIVATION, THE FOLLOWING METHOD WILL BE USED.

THERMAL GRADIENT STRESSES FROM PAGE 6 OF REF. [1] WILL BE MULTIPLIED BY THE RATIO, 1.07. PRESSURE STRESS WILL APPROPRIATELY ADJUSTED.

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REV

THERMAL DISCONTINUITY

THE THERMAL DISCONTINUITY EFFECTS WILL ALSO BE LOWER WITH THE LOWER FILM COEFFICIENTS BEING USED. THE AXIAL AT WOULD BE SMALLER. SINCE, THE LONGITUDINAL AND HOOP STRESSES ARE ADDITIVE, AS LONG AS THE LINEAR RADIAL GRADIENT STRESSES ARE LESS, THE DISCONTINUITY STRESSES WILL BE LESS.

CUSTOMER

SUP

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HARCOCK & WILCOX
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REV

PRIMARY PLUS SECONDARY STRESS (INSIDE)

USING TEMPERATURES AND PRESSURE
FROM PAGE 6 AND STRESS FOR ITER 8.

$$\begin{aligned}\text{PRESSURE STRESS} &= (4.84 \text{ KSI}) (2300 \text{ PSI} / 2200 \text{ PSI}) \\ &= 5.06 \text{ KSI (INSIDE)}\end{aligned}$$

$$\text{THERMAL EXPANSION} = 5.1 \text{ KSI (INSIDE)}$$

$$\text{THERMAL STRESS} = 78.8 (1.07) = 84.32 \text{ KSI}$$

$$\text{TOTAL STRESS} = 5.06 + 5.1 + 84.32 = 94.48 \text{ KSI}$$

ITERATION 6333 (MAXIMUM THERMAL STRESS)

$$\begin{aligned}\text{PRESSURE STRESS} &= (1.54 \text{ KSI}) (1800 \text{ PSI} / 700 \text{ PSI}) \\ &= 3.96 \text{ KSI (INSIDE)}\end{aligned}$$

$$\text{THERMAL EXPANSION} = 5.1 \text{ KSI (INSIDE)}$$

$$\begin{aligned}\text{THERMAL STRESS} &= (-2.7 \text{ KSI}) (565 / 500) \\ &= -8.7 \text{ KSI (INSIDE)}\end{aligned}$$

$$\text{TOTAL STRESS} = 3.96 + 5.1 - 8.7 = 0.36 \text{ KSI}$$

MAXIMUM PRIMARY + SECONDARY STRESS
INTENSITY RANGE

$$\sigma_{\text{P+S}} = 94.48 + 20.35 = 94.48 + 20.0 = 114.48 \text{ KSI}$$

$$114.48 \text{ KSI} > 3S_m = 51.2 \text{ KSI}$$

CUSTOMER	PROP. NO.	FILE NO.
SUBJECT	DWG. NO.	COMP. NO.
CONT. NO. 620 00	GROUP NO.	
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ALM	10/6/80	

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SERIAL NUMBER00
REVPRIMARY PLUS SECONDARY STRESS (OUTSIDE)ITER 8

$$\text{PRESSURE STRESS} = 1.32 (2300/2200) = 1.33 \text{ KSI}$$

$$\text{THERMAL EXPANSION} = 7.3 \text{ KSI}$$

$$\text{THERMAL GRADIENT STRESS} = -71.9 (1.07) = -76.93 \text{ KSI}$$

$$\text{TOTAL STRESS} = -68.3 \text{ KSI}$$

ITER 6338

$$\text{PRESSURE STRESS} = 0.42 (1200/700) = 1.08 \text{ KSI}$$

$$\text{THERMAL EXPANSION} = 7.3 \text{ KSI}$$

$$\text{THERMAL GRADIENT STRESS} = 7.0 (565/500) = 7.91 \text{ KSI}$$

$$\text{TOTAL STRESS} = 1.08 + 7.30 + 7.91 = 16.3 \text{ KSI}$$

MAXIMUM P+S STRESS INTENSITY RANGE

$$\sigma_{p+s} = -68.3 + 16.3 + 25.1 = 112.7 \text{ KSI}$$

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REV

PEAK STRESS INTENSITY (INSIDE)

ITER B (STRESS FROM PG. 12 OF REV 00)

$$\sigma = 5.5 (2300/2200) + 6.5 + 113.3 (1.07)$$

$$\sigma = 133.48 \text{ KSI (INSIDE)}$$

ITER 633B

$$\sigma = 1.8 (1500/700) + 6.5 - 9.6 (565/500)$$

$$\sigma = -0.3 \text{ KSI (INSIDE)}$$

TOTAL STRESS INTENSITY RANGE

$$S_{L-R} = 133.48 + 0.3 = 133.8 \text{ KSI}$$

$$\begin{aligned} \sigma_{\text{RANGE}} &= 133.8 \text{ KSI} + 20\% = 133.8 + 25.5 \\ &= 159.3 \text{ KSI} \end{aligned}$$

NOTE: OBE STRESS INTENSITY RANGES WERE ALSO ADDED TO THE ANALYSIS OF 70 ADDITIONAL CYCLES. THIS PROVIDES ADDITIONAL CONSERVATISM TO THE ANALYSIS.

CUSTOMER

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RABCOCK & WILCOX GENERAL CALCULATIONS

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ELASTIC-PLASTIC ANALYSIS FOR THE 70 ADDITIONAL NPI TRANSIENTS

FOLLOWING THE PROCEDURE OF REF [5],
 PARAGRAPH F-105.2.7, THE FINAL PEAK
 ALTERNATING STRESS INTENSITY, σ_{ALT} , IS:

$$\sigma_{ALT} = \frac{1}{2} K_f K_E S_{rij}^{(n)}$$

$$K_f = K_t + A(K_t - 1)$$

$$K_t = \sigma_{PEAK} / \sigma_{rms} = 159.3 / 114.5 = 1.39$$

$$A = 0.7 \quad (\text{PG 17, REF. [1]})$$

$$K_f = 1.39 + 0.7(1.39 - 1.0) = 1.65$$

K_E IS FROM FIG. F-105(a), REF. [5]

$$\frac{S_u}{3S_m} = \frac{114.5}{51.3} = 2.23$$

$$|Q_m| = \frac{(94.48 + 10.0) + (-68.3 + 10.0)}{2} - \frac{(0.36 - 10.0) + (16.3 - 10.0)}{2}$$

$$|Q_m| = 23.09 - (-1.67) = 24.76 \text{ KSI}$$

$$Q_b = (104.48 - 23.09) - [-9.64 - (-1.67)]$$

$$Q_b = 81.39 + 7.97 = 89.36 \text{ KSI}$$

$$\frac{Q_m}{Q_m + Q_b} = \frac{24.76}{24.76 + 89.36} = 0.22$$

$$\therefore K_E = 1.75$$

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REV.

$$\sigma_{ALT} = \frac{1}{2} (1.65)(1.75)(114.48) = 165.3 \text{ KSI}$$

$$\text{ALLOWABLE CYCLES} = 270 \left\{ \begin{array}{l} \text{PER REF. [5]} \\ \text{FIG. F-106(b)} \end{array} \right\}$$

$$U_A = 70/270 = 0.26$$

CUMULATIVE USAGE FACTOR (U_{TOTAL})

USAGE FACTOR FOR ALL MPI TRANSIENTS
EXCEPT THE 70 ADDITIONAL CYCLES
IS 0.441 PER PAGE 23, REF. [1].
THEREFORE, THE TOTAL USAGE FACTOR
FOR MPI NOZZLE CONSIDERING THE
70 ADDITIONAL MPI CYCLES DESCRIBED
IN REF. [7] IS;

$$U_{TOTAL} = 0.441 + 0.26 = \underline{\underline{0.70}}$$

$$= 0.70 < 1.0$$

$\therefore$ ACCEPTABLE

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CALC. BY RRS

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GROUP NO.

FILE NO.

COMP. NO.

SHEET NO. 16-113

DATE 10/8/80

BABCOCK & WILCOX
GENERAL CALCULATIONS32
DOC 1.01121811
SERIAL NUMBERREFERENCES

1. CDS 32-1119809-00, "HPI NOZZLE USAGE FACTOR."
2. REACTOR COOLANT PRIMARY PIPING STRESS REPORT, B&W CONTRACT NO. 620-0011-50, DESIGN REPORT NO. 5, "HPI NOZZLE ANALYSIS."
3. REACTOR COOLANT SYSTEM FUNCTIONAL SPECIFICATION CS(F)3-92/NSS-11/0672.
4. SPR #13-11-361-00, B&W CONTRACT NO. 620-0011.
5. USAS B31.7 DRAFT EDITION, DATED FEBRUARY 1968 WITH JUNE 1968 ERRATA.
6. B&W COMPUTER PROGRAM P91232, VERSION 1.0, "ONE DIMENSIONAL IMPLICIT TRANSIENT TEMPERATURE DISTRIBUTION."
7. FCA NO. 04-3370-00, B&W CONTRACT 620-0011-50.
8. MICROFICHE, SNAPELGE, SCHAEFER, RB, 8/27/80. (ATTACHED)

CUSTOMER

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CONT NO 620 00

CALC BY

RRS

DATE 10/6/82

PROP NO

DWG NO

GROUP NO

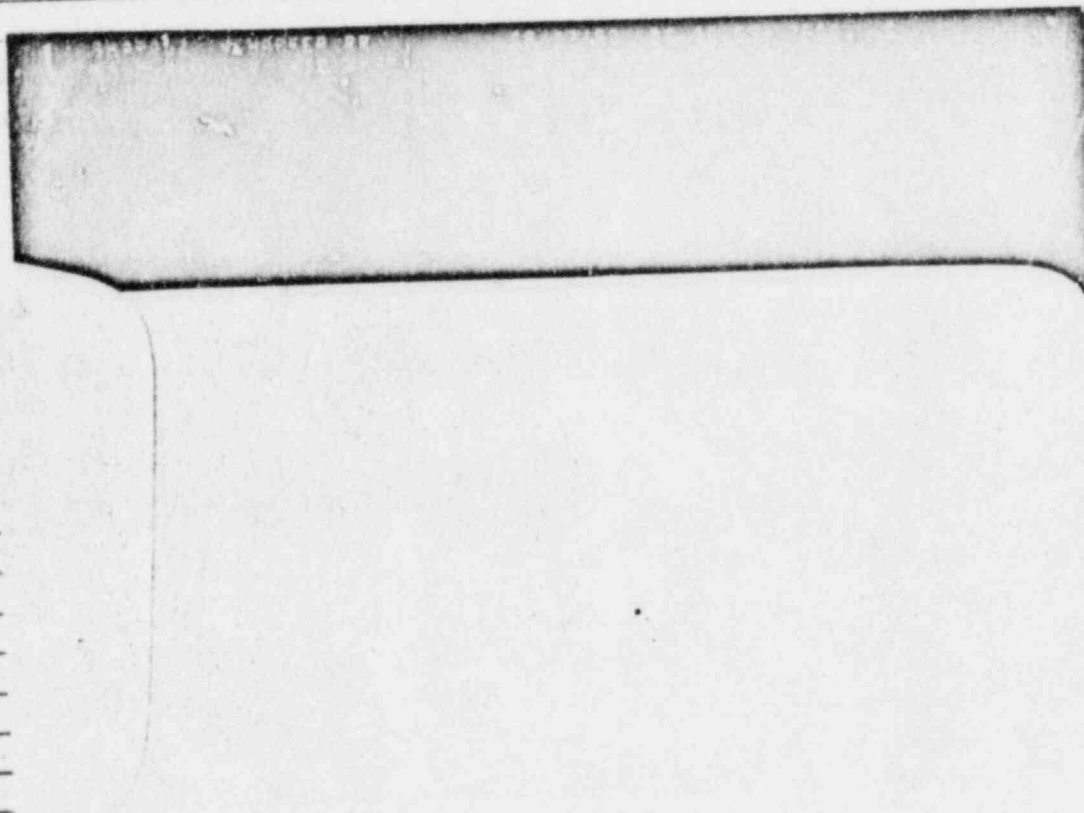
FILE NO

COMP NO

SHEET NO

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DOC ID1121811
SERIAL NUMBER00
REVREFERENCES (ATTACHED)

FICNE SNAFQGE, DATED 8/27/82.



CUSTOMER	PROP NO	FILE NO
SUBJECT	DWG NO	COMP NO
CONT NO 620.00	GROUP NO	
CALC BY RRS	DATE 8/27/82	SHEET NO 18

CALCULATION DATA/TRANSMITTAL SHEET

DOCUMENT IDENTIFIER
 CALC. 32 - 1119809 - 01
 TRANS. 86 - _____ - _____

TYPE: ☐ RESEARCH & DEVELOPMENT ☐ SAFETY ANALYSIS REPORT ☐ MUC. SERV. INPUT ☐ DESIGN RQMT. ☒ DESIGN VERIF.
☐ OTHER

TITLE HPI Nozzle Usage Factor

PREPARED BY RR Schaefer REVIEWED BY Alvin D. McKim
 TITLE Senior Engineer DATE 10/9/80 TITLE Supervisory Engineer DATE 10/9/80

PURPOSE:

This calculation is being revised to show that it is not the controlling stress document for SPR #13-11361-00 anymore.

This document is now amended by CDS #32-1121811 -00

SUMMARY OF RESULTS (INCLUDE DOC. ID'S OF PREVIOUS TRANSMITTALS & SOURCE CALCULATIONAL PACKAGES FOR THIS TRANSMITTAL)

See CDS-32-1121811-00 and CDS-32-1119809-00.

DISTRIBUTION

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CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI USAGE FACTOR (NOZZLE)	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY ADM DATE 6/1/80 CHK'D BY RB DATE 6/24/80		SHEET NO 2 of 25

PURPOSE & INTRODUCTION

B&W Size Problem Report # 13-11-361-00 described a problem at SMUD where the operators have been starting a second makeup pump following each reactor trip to obtain additional make up flow to prevent loss of indicated pressure level during the transient. All four of the HPI nozzles have been used for these occurrences, and the three nozzle which do not have continuous makeup flow to the system receive a thermal shock from the cold BWST water. The B&W Reactor System Functional Specification defines the operational events for the HPI

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
ALC BY ALM DATE 6/2/85 CHKD BY R2 DATE 6/24/85		SHEET NO 3 of 25

GENERAL CALCULATIONS

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nozzles which include 40 cycles of HPI actuations and 40 cycles of HPI testing. At the current rate of usage, SMUD will exceed their allowable lifetime cycles before the end of the 40 year life.

The purpose of this document is to calculate the fatigue usage factor for the SMUD HPI nozzles utilizing the contractual elastic-plastic fatigue analysis in reference #1. This analysis method is not as restrictive as the method utilized in the Stress Report, reference #3. With the present usage rate at SMUD, it is important to eliminate the Stress Report conservatism and justify the maximum number of allowable cycles the HPI nozzles can withstand in an attempt

CUSTOMER SMUD	PROP NO	CONT NO 620-00
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY ADM DATE 6/2/80 CHKD BY RZ DATE 6/24/80		SHEET NO 4 of 25

GENERAL CALCULATIONS

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to justify the nozzle for their lifetime cycles.

CONCLUSION & RESULTS

MAXIMUM PRIMARY + SECONDARY STRESS RANGE = 128.9 ksi
 $> 3 S_m$; therefore an elastic-plastic fatigue analysis was performed.

Usage Factor = 0.441 < 1.0 for 40 test transients
 and 40 rapid depressurization transient

Maximum number of rapid depressurization cycles
 is 110 in conjunction with 40 test transient cycles.

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY ADM DATE 6/2/80 CHKD BY JBS DATE 6/24/80	SHEET NO 5 of 5	

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GENERAL CALCULATIONS

PRIMARY + SECONDARY STRESS INTENSITIES
SEGMENT # 8 (L-R INTENSITY)

Iteration	Pressure (psi)	Pressure Stress (ksi)		Thermal Stress (ksi)		Stress Report Thermal Stress (ksi)		Total Stress Intensity (ksi)	
		Ins.	Out.	Inside	Outside	Inside	Outside	Inside	Outside
3	2200	4.84	1.32	5.1	7.3	63.7	-58.2	73.64	-49.58
6	2200	4.84	1.32	5.1	7.3	77.5	-70.8	87.44	-62.18
8	2200	4.84	1.32	5.1	7.3	76.8	-71.9	88.74	-63.28
15	2200	4.84	1.32	5.1	7.3	69.3	-63.0	79.24	-54.38
30	2200	4.84	1.32	5.1	7.3	7.7	-6.9	17.64	1.72
1211	2200	4.84	1.32	5.1	7.3	-0.4	0.3	9.54	8.42
1491	2200	4.84	1.32	5.1	7.3	0.0	0.0	9.94	8.62
42021	2200	4.84	1.32	5.1	7.3	0.0	0.0	9.24	9.22
27967	2200	4.84	1.32	5.1	7.3	-0.7	0.6	9.94	-69.68
5072	2200	4.84	1.32	5.1	7.3	86.0	-78.3	95.94	-58.18
5060	2200	4.84	1.32	5.1	7.3	73.1	-66.8	83.04	-50.68
5063	2200	4.84	1.32	5.1	7.3	97.9	-85.3	107.84	-21.14
5124	2100	4.62	1.26	5.1	7.3	33.0	-29.7	42.72	-2.02
5237	1800	3.96	1.08	5.1	7.3	11.5	-10.4	20.56	7.50
6059	1000	2.2	0.6	5.1	7.3	0.4	-0.4	7.26	8.18
6367	800	1.76	0.48	5.1	7.3	0.4	7.0	-1.06	14.72
6336	700	1.54	0.42	5.1	7.3	-7.7	3.1	2.92	10.76
6850	600	1.32	0.36	5.1	7.3	-3.5	0.0	± 19.9	± 28.1
1708E	0.0	0.0	0.0	0.0	0.0	± 0.0	0.0	± 10.0	± 14.0
OBE	0.0	0.0	0.0	0.0	0.0	0.0	0.0	± 10.0	± 14.0

See the following sheet for references and table explanation.

CUSTOMER SMUD

SUBJECT HPI NOZZLE USAGE FACTOR

CALC BY ADM DATE 6/2/80 CHKD BY RB DATE 6/24/80

PROP NO

DWG NO

COMP NO 50

CONT NO 620-000

FILE NO

GROUP NO

SHEET NO 6

GENERAL CALCULATIONS

32-1119809 01

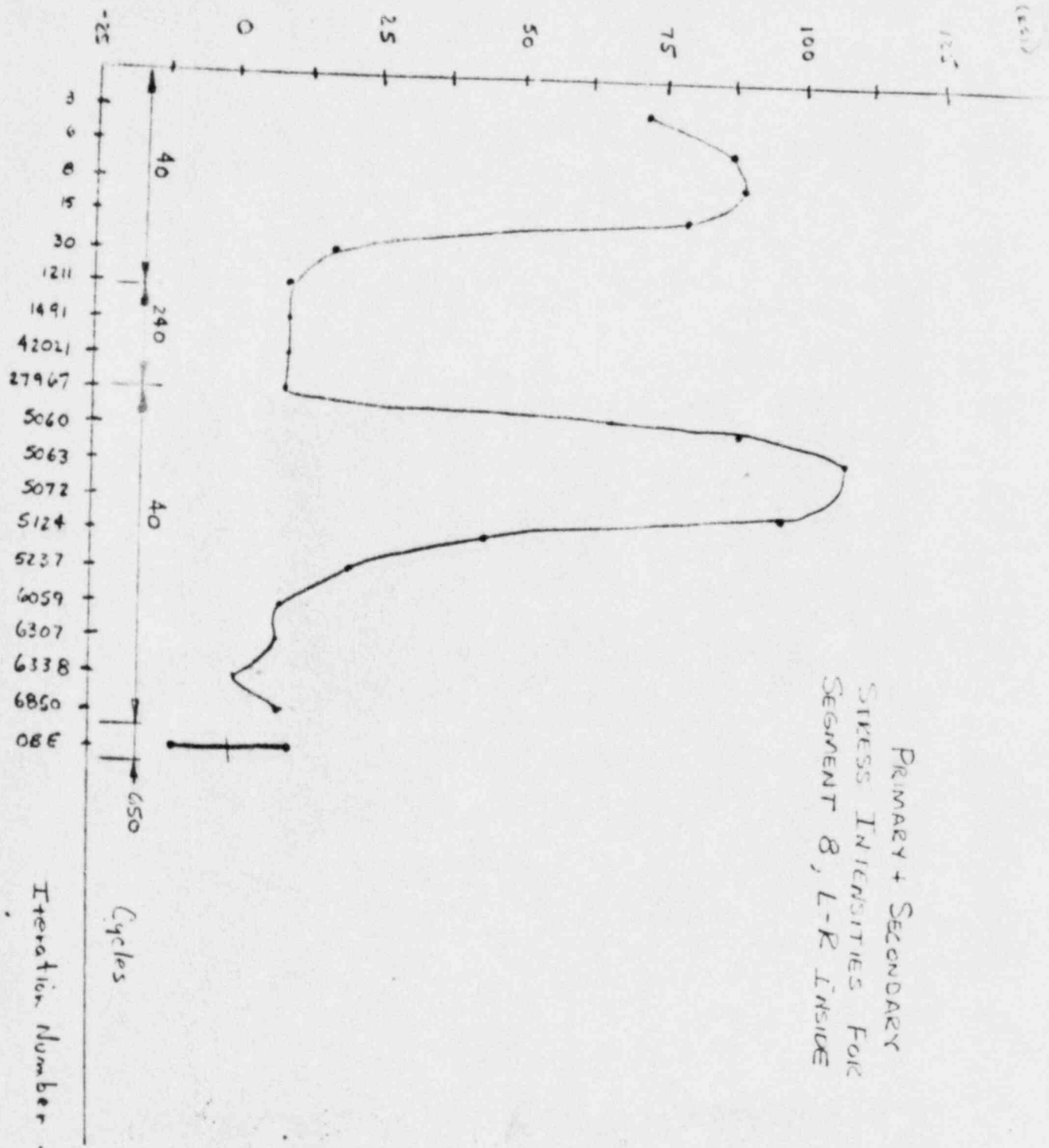
TABLE REFERENCES AND EXPLANATIONS

- Pressure stresses are obtained by multiplying the pressure stresses from reference #3, page 5-17 (Iteration #1) by the ratio of actual pressure / 1000 psi (pressure used in reference #3).
- Thermal expansion stresses are from reference #3, page 1-6 for section 2.
- Stress Report thermal stresses are from reference #3, page 5-17.
- The cross section position that experiences positive thermal expansion stresses is used to maximize the inside intensity which is the critical intensity.
- 2X OBE stresses are from reference #3, page 1-6 for section 2. This is the stress state range for a cold earthquake.
- Hot earthquake stresses and cycles will be applied and analyzed for in the primary + secondary and peak stress intensity ranges sheets in this document. OBE stresses are from reference #3, page 1-6. OBE stresses = OBE + Thermal - Thermal.
- A graph of the L-R inside intensities is shown on sheet 8 to aid in determining ranges for maximum and minimum intensity values.

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY ALM DATE 6/2/80 CHK'D BY RB DATE 6/24/80		SHEET NO 7

GENERAL CALCULATIONS

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CUSTOMER	SMUD	PROP NO	CONT NO
SUBJECT		DWG NO	FILE NO
		COMP NO	GROUP NO
CALC BY	AM	DATE	6/24/80
CHK'D BY	AS	DATE	6/24/80
		SHEET NO	8 of 8

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GENERAL CALCULATIONS

DETERMINATION OF THE NUMBER OF STRESS
CYCLES THE PRIMARY + SECONDARY STRESS
INTENSITY RANGE EXCEEDS $3S_m$

The primary + secondary stress intensity range limit per reference #1, paragraph F-104.4, is $3S_m$. The $3S_m$ value at the critical location, segment #8, is 51.3 ksi which is exceeded for a number of stress intensity cycles. Since the $3S_m$ limit is exceeded, an elastic-plastic fatigue analysis must be performed in accordance with reference #1, paragraph F-105.2.7. This fatigue method is valid only if the number of cycles that exceed $3S_m$ are less than 250. The number of cycles that exceed $3S_m$ are determined on sheets 10 and 11.

CUSTOMER SMUD	PROP NO	CONT NO
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO 620-0011
	COMP NO 50	GROUP NO
CALC. BY AJM DATE 6/2/80	CHKD BY RD DATE 6/24/80	SHEET NO 9 of 25

GENERAL CALCULATIONS

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MAXIMUM PRIMARY + SECONDARY STRESS
INTENSITY RANGE

Range is comprised of $|(\text{Iteration } 6338 + \overset{-S_{gn}}{OBE})| +$
 $(\text{Iteration } 5063 + \overset{+S_{gn}}{OBE})$ and can occur for
40 cycles.

$$\sigma_{\text{Range}}^{\text{Pri} + \text{Sec}} = 1.06 + 10.0 + 107.84 + 10.0 = 128.9 \text{ ksi} > 3S_m$$

2nd MAXIMUM PRIMARY + SECONDARY STRESS
INTENSITY RANGE

Range is comprised of $|0 \text{ stress state} + \overset{-S_{gn}}{OBE}| +$
 $(\text{Iteration } 8 + \overset{+S_{gn}}{OBE})$ and can occur for 40 cycles

$$\sigma_{\text{Range}}^{\text{Pri} + \text{Sec}} = 0 + 10.0 + 88.74 + 10.0 = 108.74 \text{ ksi} > 3S_m$$

3rd MAXIMUM PRIMARY + SECONDARY STRESS
INTENSITY RANGE

Range is comprised of $|0 \text{ stress state} + \overset{-S_{gn}}{OBE}| +$
 $(\text{Iteration } 42021 + \overset{+S_{gn}}{OBE})$ and can occur for 40 cycles

$$\sigma_{\text{Range}}^{\text{Pri} + \text{Sec}} = 0 + 10.0 + 9.94 + 10.0 = 39.94 \text{ ksi} > 3S_m$$

CUSTOMER	SMUD	PROP NO	CONT NO 620-001
SUBJECT	HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
		COMP NO 50	GROUP NO
CALC BY 111M	DATE 6/2/80	CHK'D BY RB	DATE 6/24/80
			SHEET NO 10 of 40

GENERAL CALCULATIONS

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Remaining stress intensity ranges are less than $3S_m$.

Total number of cycles in which the primary + secondary stress intensity range exceeds $3S_m$ is 80; therefore, the elastic-plastic fatigue analysis is valid.

CUSTOMER	SMUD	PROP NO	CONT NO 620-001
SUBJECT	HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
		COMP NO 50	GROUP NO
CALC BY	AFM	DATE 6/19/80	CHKD BY AS
		DATE 6/24/80	SHEET NO 11 of 25

GENERAL CALCULATIONS

32-1119809 001

PEAK STRESS INTENSITIES SEGMENT
8 L-R INSIDE

Iteration	Pressure Psi	Pressure Stress Ksi	Thermal Exp. T (°F)	Stress Rpt. Therm T (°F)	Total Stress (ksi)
3	2200	5.5	6.5	109.0	121.0
6	2200	5.5	6.5	115.0	127.0
8	2200	5.5	6.5	113.3	125.3
15	2200	5.5	6.5	95.0	107.0
30	2200	5.5	6.5	8.7	20.7
1211	2200	5.5	6.5	-0.4	11.6
1491	2200	5.5	6.5	0.0	12.0
42021	2200	5.5	6.5	0.0	12.0
17967	2200	5.5	6.5	-0.8	11.2
5072	2200	5.5	6.5	117.3	129.3
5060	2200	5.5	6.5	147.4	159.4
5063	2200	5.5	6.5	143.3	155.3
5124	2100	5.3	6.5	39.9	51.7
5237	1900	4.5	6.5	14.2	25.2
5059	1000	2.5	6.5	0.5	9.5
6307	800	2.0	6.5	0.5	9.0
6338	700	1.8	6.5	-9.6	-1.3
6850	600	1.5	6.5	-1.3	6.7
2XOLE	0.0	0.0	0.0	0.0	25.5
OBE	0.0	0.0	0.0	0.0	±12.7

See the following sheet for references
and table explanations.

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY KPM DATE 6/1/83	CHKD BY RB	DATE 6/24/83
		SHEET NO 12 of 25

GENERAL CALCULATIONS

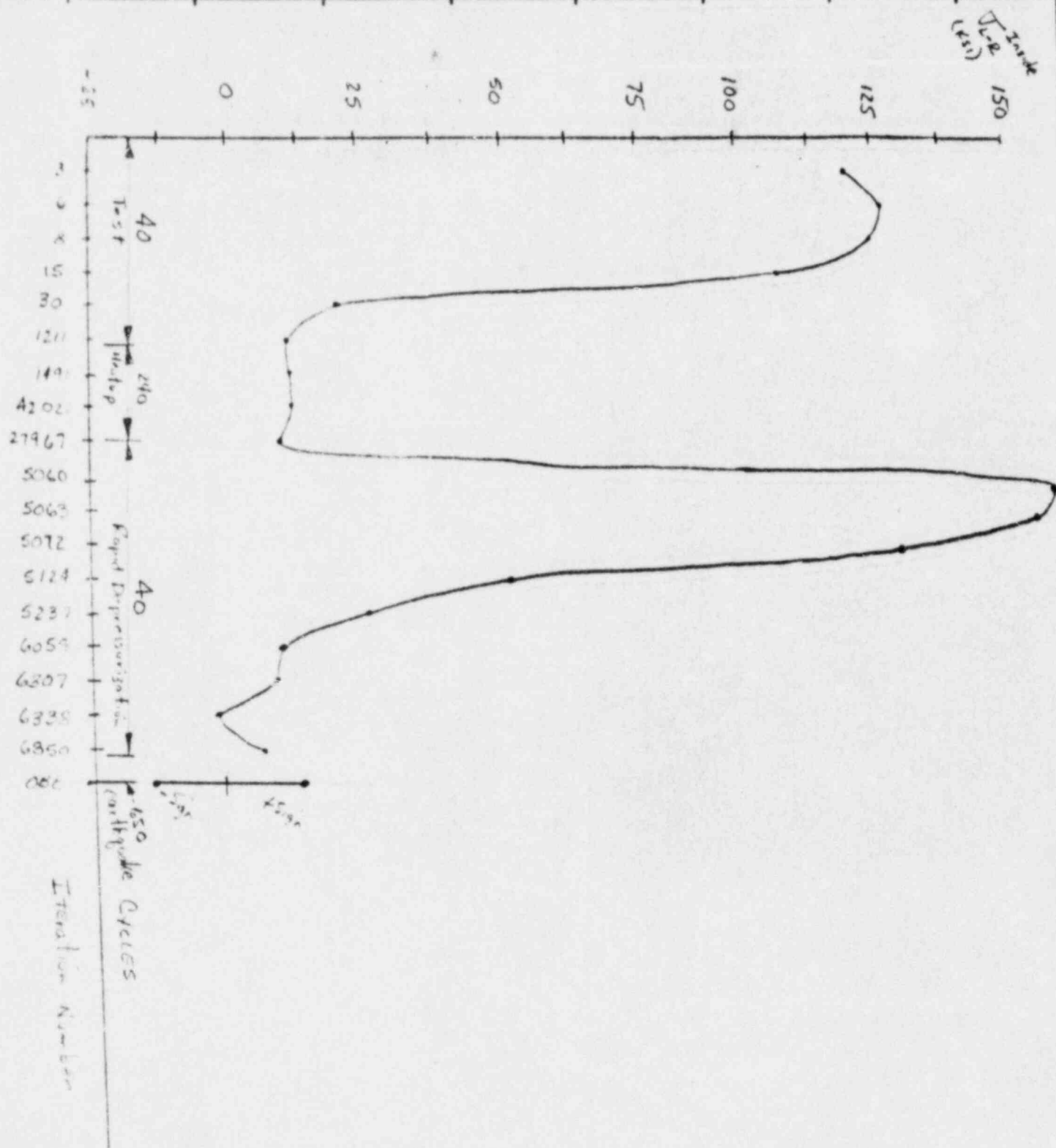
32-1119809 001

TABLE REFERENCES AND EXPLANATIONS

- Pressure stresses are obtained by multiplying the peak pressure from reference #3, page 5-17 (Iteration #1) by the ratio of actual pressure divided by 1000 psi (pressure used in reference #3)
- Thermal expansion stresses are from reference #3, page 1-6 for section 2 multiplied time the bending stress concentration factor from reference #3, page 1-10.
- Stress Report thermal stresses are from ref. #3, page 5-17.
- The cross sectional position that experiences positive thermal expansion stresses is used to maximize the critical inside intensity.
- SLOBE stresses are from reference #3, page 1-6 for section 2 multiplied times the bending stress concentration factor from reference #3, page 1-10. This is the range for a cold earthquake.
- A graph of the L-R inside peak stress intensities is shown on the following sheet to aid in determining ranges for maximum and minimum intensity values.

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE LEAKAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY N/A DATE 6/1/85	CHEK BY RB DATE 6/24/80	SHEET NO 13 of 25

GENERAL CALCULATIONS



CUSTOMER SMOD

PROP NO

CONT NO

SUBJECT

DWG NO

FILE NO

COMP NO

GROUP NO

CALC BY J. M. DATE 6/2/80 CHKD BY R3 DATE 6/24/80

SHEET NO 14 of 25

GENERAL CALCULATIONS

FATIGUE ANALYSIS

The fatigue analysis is performed in accordance with paragraph F-105.2.7 of reference # 1, "Simplified Elastic - Plastic Discontinuity Analysis". The primary plus secondary stress intensity ranges determined on sheets 10 and 11 shows that the primary plus secondary ranges do not exceed 35_{sm} for more than 250 cycles; therefore, this method of fatigue evaluation is validated.

The peak stress intensities and cycles are presented in graphic form on sheet 14. This is used to determine maximum peak stress intensity ranges and cycles used in the usage factor calculations. Actual peak stress stress intensity value can be obtained from the tabulation on sheet 12. Primary plus secondary intensities are also used in the fatigue analysis:

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC BY NDM DATE 6/7/80 CHK'D BY NB DATE 6/24/80		SHEET NO 15 of 25

GENERAL CALCULATIONS

32-1119809 0/1

These values are obtained from the tabulation on sheet 6.

MAXIMUM PEAK STRESS INTENSITY RANGE

Range is comprised of (Iteration 5060 + ^{+sign}OBE) + (Iteration 6338 + ^{-sign}OBE) and can occur for 40 cycles.

$$\sigma_{\text{peak}}^{\text{range}} = 159.4 + 1.3 + 25.5 = 186.2 \text{ ksi}$$

The primary + secondary range associated with these transients = 128.9 ksi > $3S_m = 51.3 \text{ ksi}$ (see sht. 10, 2xOBE + iterations 6338 and 5063); therefore, an elastic plastic analysis must be performed for these 40 cycles. Following the procedure of reference #1, paragraph F-105.2.7, the final Peak Alternating Stress Intensity, σ_{alt} , is:

$$\sigma_{\text{alt}} = \frac{1}{2} K_f K_e S_{ij}^{(n)}$$

CUSTOMER SMUD	PROP NO	CONT NO 320-001
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
CALC. BY ADM DATE 6/2/80 CHK'D BY RB DATE 6/24/80		SHEET NO 16 of 2

GENERAL CALCULATIONS

where: $K_p = K_t + A(K_t - 1)$

$$K_t = \frac{S_{rij}^{(p)}}{S_{rij}^{(n)}} = \frac{\text{Peak Stress Intensity Range}}{\text{Primary + Secondary Intensity Range}}$$

$$= 186.2 / 128.9 = 1.44$$

$A = 0.7$ for Stainless Steel, from Figure D-201, reference #1

$$\therefore K_p = 1.44 + 0.7(1.44 - 1.0) = 1.798$$

K_E is determined from Figure F-45(a). Below is the calculation of the parameters necessary to obtain K_E :

$$\frac{S_n}{3S_m} = \frac{\text{Primary + Secondary Intensity Range}}{3S_m}$$

$$\frac{S_n}{3S_m} = 1289 / 51.3 = 2.51$$

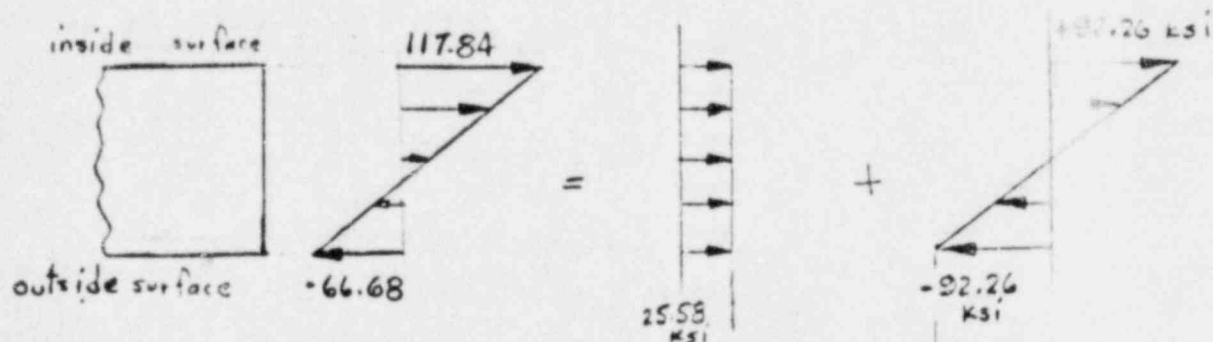
$|Q_m| / [|Q_m| + |Q_b|]$ is determined on the next sheet.

CUSTOMER SMUD	PROP NO	CONF NO 620-001
SUBJECT HPI NOZZLE USAGE FACTOR	DWG. NO	FILE NO
	COMP NO 50	GROUP NO
CALC. BY ADM DATE 6/2/80	CHKD BY R8	DATE 6/24/80
		SHEET NO 17 of 18

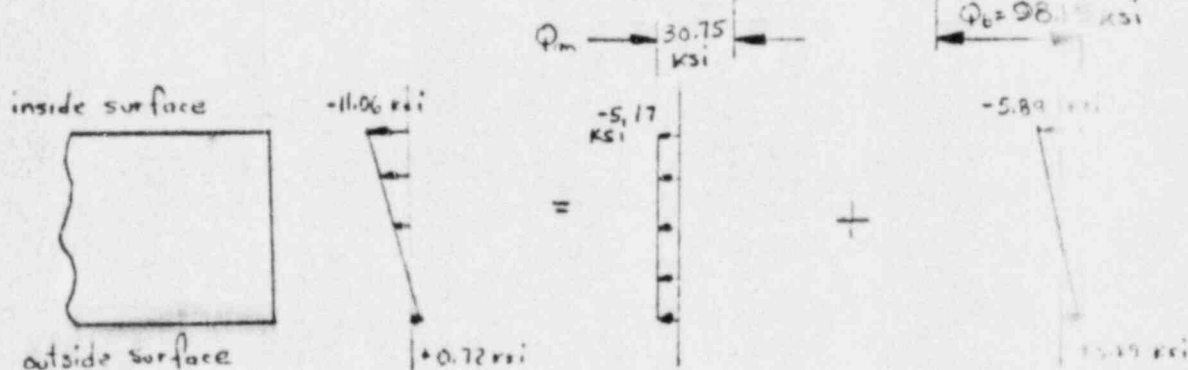
GENERAL CALCULATIONS

32-1119800 001

(Condition 1a = Iteration 5063 + OBE)



(Condition 1b = Iteration 6338 - sign OBE)



$$\frac{Q_m}{Q_m + Q_b} = \frac{30.75}{30.75 + 98.15} = 0.239$$

$$\therefore K_E = 1.95$$

from Figure E-105
reference #1

CUSTOMER	SMUD	PROP NO	CONF NO 620-00
SUBJECT	HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
		COMP NO 50	GROUP NO
CALC BY ADM	DATE 6/2/80	CHKD BY RS	DATE 6/24/80
			SHEET NO 18 of

GENERAL CALCULATIONS

32-1119309 001

$$\sigma_{alt} = \frac{1}{2} K_f K_e S_{e1}^{(n)}$$

$$\sigma_{alt} = (\frac{1}{2})(1.748)(1.95)(128.9) = 220.0 \text{ ksi}$$

Allowable cycles = 130 from reference #1, Figure H-106(b)

$$\text{Usage Factor} = U_1 = 40/130 = 0.308$$

2ND MAXIMUM PEAK STRESS INTENSITY RANGE

Range is comprised of (Iteration 6 + ^{+sign}OBE) + |(0 stress + ^{+sign}OBE)| and can occur for 40 cycles.

$$\sigma_{peak}^{range} = (12.7 + 12.7) + |(0 - 12.7)| = 152.4 \text{ ksi}$$

The primary + secondary range associated with these transients = $(88.74 + 10) + |(0 - 10.0)| = 108.74^* \text{ ksi}$

> $3S_m = 51.3 \text{ ksi}$; therefore, an elastic-plastic analysis must be performed for these 40 cycles. Following

* Iteration 8 + OBE + 0 stress state + OBE (see sheet 6)

CUSTOMER	SMUD	PROP NO	CONT NO	620-001
SUBJECT	HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO	
		COMP NO	50	GROUP NO
CALC BY	ADM	DATE	6/2/80	CHK'D BY
		DATE	6/24/80	
				SHEET NO 19 of 1

GENERAL CALCULATIONS

32-1119809 001

the procedure of reference #1, paragraph F-105.2.7,
the final Peak Alternating Stress intensity, σ_{alt} ,
is :

$$\sigma_{alt} = \frac{1}{2} K_f K_e S_{rij}^{(n)}$$

$$\text{where: } K_f = K_t + A(K_t - 1.0)$$

$$K_t = \frac{S_{rij}^{(p)}}{S_{rij}^{(n)}} = \frac{\text{Peak Stress Intensity Range}}{\text{Primary + Secondary Intensity Range}}$$

$$= 152.4 / 108.74 = 1.40$$

$A = 0.7$ for Stainless Steel, from
Figure D-201, reference #1

$$\therefore K_f = 1.40 + 0.7(1.40 - 1.0) = 1.68$$

K_e is determined from Figure F-105(a).
Below is the calculation of the parameter
necessary to obtain K_e :

$$S_n / 3S_m = \frac{\text{Primary + Secondary Intensity Range}}{3S_m}$$

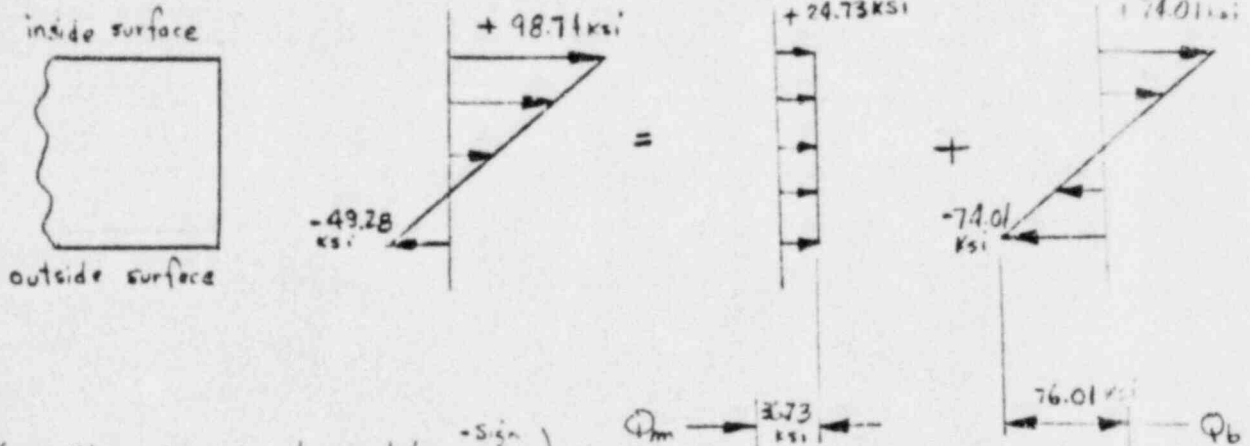
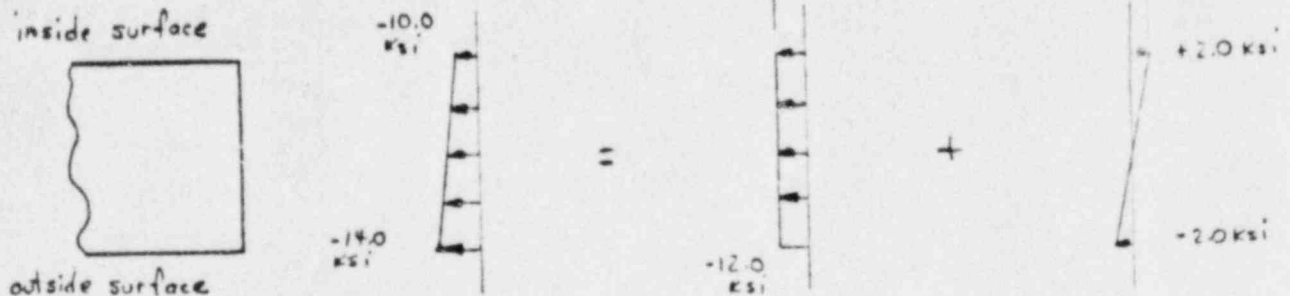
$$S_n / 3S_m = 108.74 / 51.3 = 2.12$$

$|Q_m| / [|Q_m| + |Q_b|]$ is determined on
the next sheet.

CUSTOMER SMUD	PROP NO	CONT NO 620-1
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
	COMP NO 50	GROUP NO
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GENERAL CALCULATIONS

32-1119809 0/1

(Condition 2a - Iteration 8 + S_{jr} OBE)

(Condition 2b - 0 stress state $-S_{jr}$ OBE)


$$\frac{Q_m}{Q_m + Q_b} = \frac{36.73}{36.73 + 76.01} = 0.33$$

$$\therefore K_E = 1.7$$

from figure F-105(a)
reference #1

CUSTOMER SMUD	PROP NO	CONT NO 620-00
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
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CALC. BY ADM DATE 6/2/80	CHEK BY RD DATE 1/24/80	SHEET NO 21 of 23

GENERAL CALCULATIONS

$$\sigma_{alt} = \frac{1}{2} K_f K_e S_{rij}^{(n)}$$

$$\sigma_{alt} = \frac{1}{2} (1.68)(1.7)(108.74) = 155.3 \text{ ksi}$$

Allowable cycles = 300 from reference #1, Figure F-106(b)

$$\text{Usage Factor} = U_2 = 40/300 = 0.133$$

3rd MAXIMUM PEAK STRESS INTENSITY RANGE

Range is comprised of (Iteration 27967 ^{+sign} OBE) + |(0 stress state + ^{-sign} OBE)| and can occur for 240 cycles.

$$\sigma_{peak}^{range} = (11.2 + 12.7) + |(0 - 12.7)| = 36.6 \text{ ksi}$$

The primary + secondary range associated with these iterations = $(9.24 + 10.0) + |(0 - 10.0)| = 29.24 \text{ ksi} < 3S_m = 51.4 \text{ ksi}$; therefore, the final Peak Alternating Stress Intensity Range, σ_{alt} , is:

$$\sigma_{alt} = \sigma_{peak}^{range} / 2.0 = 36.6 / 2 = 18.3 \text{ ksi}$$

CUSTOMER SMUD	PROP NO	CONT NO 670-0011
SUBJECT HPI NOZZLE USAGE FACTOR	DWG NO	FILE NO
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GENERAL CALCULATIONS

32-1119809 0

Allowable cycles = ∞

$$\text{Usage Factor} = U_3 = 240/\infty = 0.0$$

4th MAXIMUM PEAK STRESS RANGE

This range is comprised of $\pm$ OBE and can occur for the remainder of the earthquake cycles = $650 - 40 - 40 - 240 = 330$. The peak stress range is less than case 3 range which resulted in an usage factor of 0.0; therefore, $U_4 = 0.0$

Cumulative Usage Factor = U_{TOTAL}

$$U_{\text{TOTAL}} = U_1 + U_2 + U_3 + U_4 = 0.308 + 0.133 + 0.0 + 0.0$$

$$U_{\text{TOTAL}} = 0.441 < 1.0, \text{ Acceptable}$$

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE FATIGUE ANALYSIS	DWG NO	FILE NO
	COMP NO 50	GROUP NO
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GENERAL CALCULATIONS

DETERMINE OF ALLOWABLE RAPID DEPRESSURIZATION
CYCLES ASSUMING 40 CYCLES OF TEST TRANSIENT

The usage factor for 40 cycles of test
transient = 0.133 (see sheet 22, U_2)

The remaining usage factor available for
rapid depressurization transient cycles =
 $1.0 - 0.133 = 0.867$

Each rapid depressurization transient cycle
results in an usage factor contribution
of $1/130 = 0.0077$ (see sheet 19)

Therefore, the allowable rapid depressurization
transient cycles, X , is:

$$0.867 = X (0.0077)$$

$$X = 112, \text{ say } 110$$

CUSTOMER SMUD	PROP NO	CONT NO 620-0011
SUBJECT HPI NOZZLE FATIGUE ANALYSIS	DWG NO	FILE NO
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	DATE 6/24/80	SHEET NO 24 of 25

BARCOCK & WILCOX
 GENERAL CALCULATIONS

32-1119809

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SERIAL NUMBER

REV

REFERENCES

- 1) USAS B31.7 Draft Edition dated February 1968 with June 1968 errata.
- 2) 1971 ASME Section III
- 3) Design Report #5 - Thermal Mechanical Analysis of 2 1/2" Sch 160 Make-up and High Pressure Injection Nozzle for Sacramento Utility District - 620-0011-50, Design Report revision #1 dated 1-15-73.

CUSTOMER	SMUD	PROP NO	FILE NO
PROJECT	HPI NOZZLE FATIGUE ANALYSIS	TITLE NO	TITLE NO 50
QNT NO	620 00 11	GROUP NO	
CALC BY	ADM	DATE	6/7/60
		SHEET NO	15 of 25