

— B&W 177-FA Owners Group —

**EFFECTS OF ASYMMETRIC LOCA LOADINGS**

— Phase II Analysis —

**Supplement 1**

**Responses to NRC Questions**

**Document No. 77-1126594-00**

**BAW-1621, Supp. 1**

**June 1981**

**Babcock & Wilcox**

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— B&W 177-FA Owners Group —  
EFFECTS OF ASYMMETRIC LOCA LOADINGS  
— Phase II Analysis —

Supplement 1  
Responses to NRC Questions

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## INTRODUCTION AND SCOPE

On January 25, 1978, the U. S. Nuclear Regulatory Commission's Division of Operating Reactors requested all PWR licensees to proceed with an evaluation of asymmetric loss-of-coolant accident (LOCA) loadings. In response to this request, The Babcock & Wilcox Company (B&W) and its consultant, EDS Nuclear, Incorporated, performed a detailed Phase II evaluation to determine the effects of these loadings on components, equipment, and structures for the B&W 177-Fuel Assembly Owners Group plants. The results were summarized and reported in BAW-1621 in July 1980 ("Effects of Asymmetric LOCA Loadings -- Phase II Analysis").

On January 9, 1981, Robert W. Reid, Chief of Operating Reactors, prepared a letter to all B&W licensees requesting that answers be provided to specific questions related to the Phase II report. Subsequently, on April 2 and 3, 1981, the Owners Group subcommittee, B&W, and EDS representatives met with the NRC staff and the NRC's consultant, EG&G, to review and discuss the questions and to define an acceptable format and scope for responding to each question so that all issues could be resolved with one final response.

This report provides the agreed upon responses and additional information and is submitted for the following member utilities of the B&W 177-FA Owners Group:

Arkansas Power & Light	Arkansas Nuclear One, Unit (ANO-1)
Duke Power Company	Oconee Units 1, 2, and 3
Florida Power Corporation	Crystal River 3 (CR-3)
General Public Utilities	Three Mile Island 1 and 2 (TMI-1, -2)
Sacramento Municipal Utility District	Rancho Seco
Toledo Edison Company	Davis Besse 1 (DB-1)

The questions and responses are numbered to be consistent with the numbering in the January 9, 1981, letter from R. W. Reid. The appendixes are used for the convenience of summarizing technical data.

1.1. How were friction factors obtained? (Section 4.3.2)

Response

Friction factors were calculated internally by the CRAFT2 computer code<sup>1</sup> (reference section 2.2.5) based on flow path input parameters,  $L/D_h$  (flow path length/hydraulic diameter) and  $D_h$  (hydraulic diameter). As stated in section 4.3.2 of BAW-1621<sup>2</sup>, insulation assumptions were made in such a fashion as to produce conservative (maximum resistance) values of  $L/D_h$  and  $D_h$ . An additional multiplier (greater than one) was then applied to the geometric  $L/D_h$  to account for flow over concrete.

The net result of this method is a very conservative calculational procedure for friction factors.

1.2. What  $L/D_h$  was used for turns not equal to 90 degrees? (Section 4.3.2)

Response

No flow path turns other than 90° were considered. As stated in the response to Question 1.1, frictional losses were accounted for in a very conservative manner. In addition, turning losses associated with circumferential flow paths (30°) in the reactor vessel cavity were considered negligible when compared to frictional and form losses in reference 3.\* Furthermore, most circumferential flow paths undergo some type of "gradual" area change (flow over reactor vessel inlet and outlet piping, core flood line piping and detector thimbles), which was conservatively modeled using the equations in section 4.3.2 of BAW-1621<sup>2</sup> for sudden expansions and contractions.

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\* See section 6, page 208, of that report.

1.3. Supply sample calculation of a form loss coefficient. (Section 4.3.2)

Response

Consider nodes 1 and 2, whose boundaries respectively are defined by points 12341 and 14561, respectively (shown in Figure 1.3-1). The nodes are connected by flow path ①, which extends from point A to point B (i.e., from the center of node 1 to the center of node 2). Notice that the flow path is impeded by a pipe in node 1 and by the detector thimbles in node 2.

The form loss coefficient for this flow path would be calculated using the equations for sudden expansion and contraction presented in section 4.3.2 of BAW-1621.<sup>2</sup> Letting

$\xi_1$  - flow area at point A (minimum flow area from A $\rightarrow\alpha$ , also minimum flow path area from A $\rightarrow$ B),

$\xi_2$  - flow area from  $\alpha \rightarrow \beta$ , and

$\xi_3$  - flow area from  $\beta \rightarrow B$ ,

$$K_F = \left(1 - \frac{\xi_1}{\xi_2}\right)^2 + 0.4 \left(1 - \frac{\xi_3}{\xi_2}\right) \left(\frac{\xi_1}{\xi_3}\right)^2$$

and

$$K_R = \left(1 - \frac{\xi_3}{\xi_2}\right)^2 \left(\frac{\xi_1}{\xi_3}\right) + 0.4 \left(1 - \frac{\xi_1}{\xi_2}\right)$$

where

$K_F$  = forward form loss coefficient,

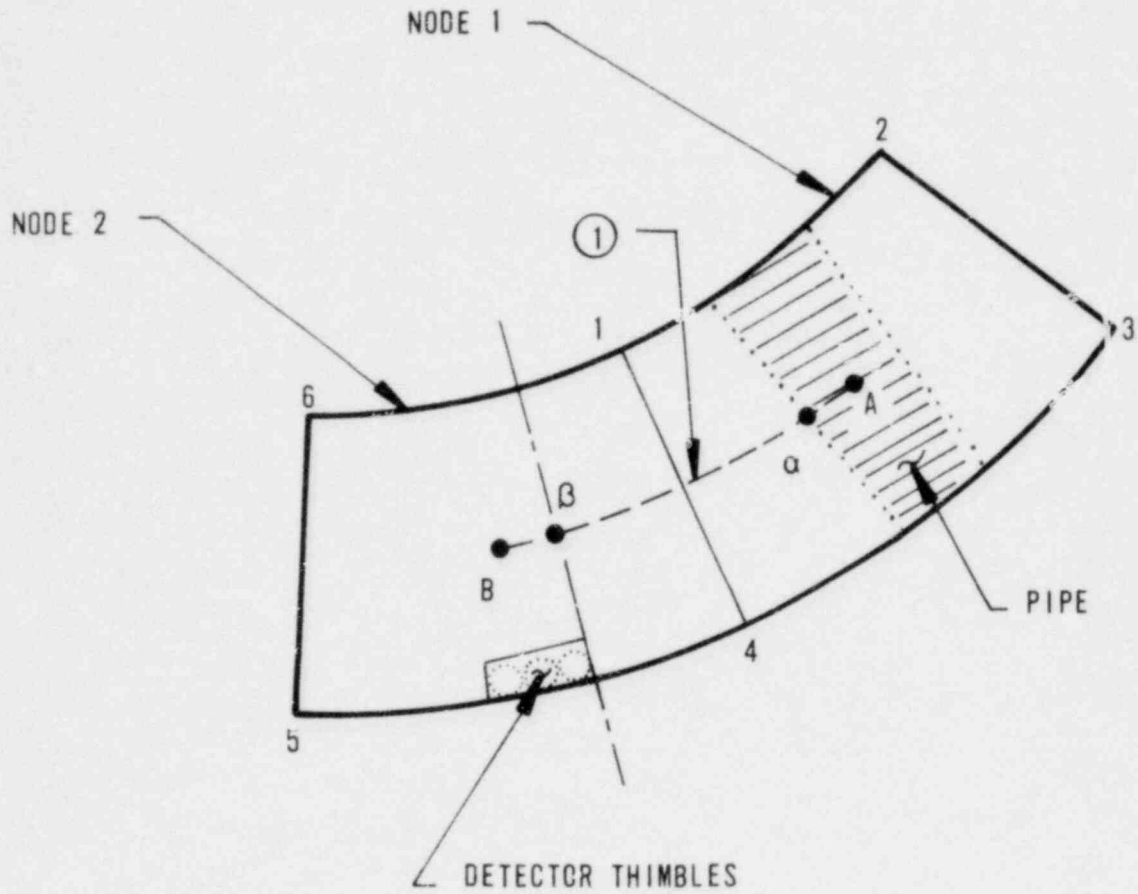
$K_R$  = reverse form loss coefficient for flow path 1 in Figure 1.3-1.

Figure 1.3-1. Form Loss Coefficient Flow Path

A - CENTER OF NODE 1

B - CENTER OF NODE 2

FLOW DIRECTION - A → B





1.4. What, if any, equipment is in the path of motion of venting devices?  
(Section 4.3.2)

Response

The service support structure and the reactor vessel canal seal plate are the only items in the path of motion of these venting devices. The canal seal plate has been removed on some plants (e.g., Oconee, TMI-1, and ANO-1).

1.5. What uncertainty factors were applied to calculated flow areas? Demonstrate that uncertainty factors were applied in a conservative manner. (Section 4.3.2)

Response

No uncertainty factors were applied to calculated flow path areas. Said areas were determined from utility-supplied, as-built drawings. Physical path areas from the drawings were then reduced by making conservative insulation assumptions and only the minimum area along any given path was used as a computer code input parameter (see section 4.3.2 of BAW-1621<sup>2</sup>).

The result of the procedure above is a conservative calculation of flow path areas.

1.6. Supply the values of any uncertainty factors that may have been applied to other parameters such as volumes, flow lengths, etc. (Section 4.3.2)

Response

No uncertainty factors were applied to such parameters as node volumes, flow path lengths, etc. Such geometric parameters were determined from utility-supplied, as-built drawings. Section 4.3.1 of BAW-1621<sup>2</sup> sets forth the node volume conservatisms employed in the reactor vessel cavity pressure calculations, while section 4.3.2 presents conservatisms introduced in the generation of geometric parameters such as flow path lengths, areas, loss coefficients, etc.

Geometric computer model input parameters were determined in a conservative fashion.

- 1.7. Supply the initial conditions (pressure and temperature) used in the mass and energy release rate calculations. Demonstrate the generic applicability of these values. (Section 4.3.2)

#### Response

Table 4.3.8<sup>2</sup> gives the initial conditions for mass and energy release rate calculations. Section 4.3.4.2 gives additional mass and energy inputs.

An initial reactor coolant system pressure of 2200 psia in the plenum volume directly above the core was used for the mass and energy release calculations. The reactor coolant system pressure drop is calculated by CRAFT2, and pressure at each node throughout the system is determined.

The loop geometry and fluid conditions of the 177-FA skirt-supported plants are very similar and have been grouped generically in other analyses<sup>4,5</sup>. In order to maximize mass and energy release, 102% of the release from the highest power level plant was used. The response to question 2.10 presents a comparison of plant parameters.

- 1.8. A drag coefficient of zero does seem unrealistic; however, justification should be provided for the empirical mass multiplier of 2.0. (Section 4.3.2)

Response

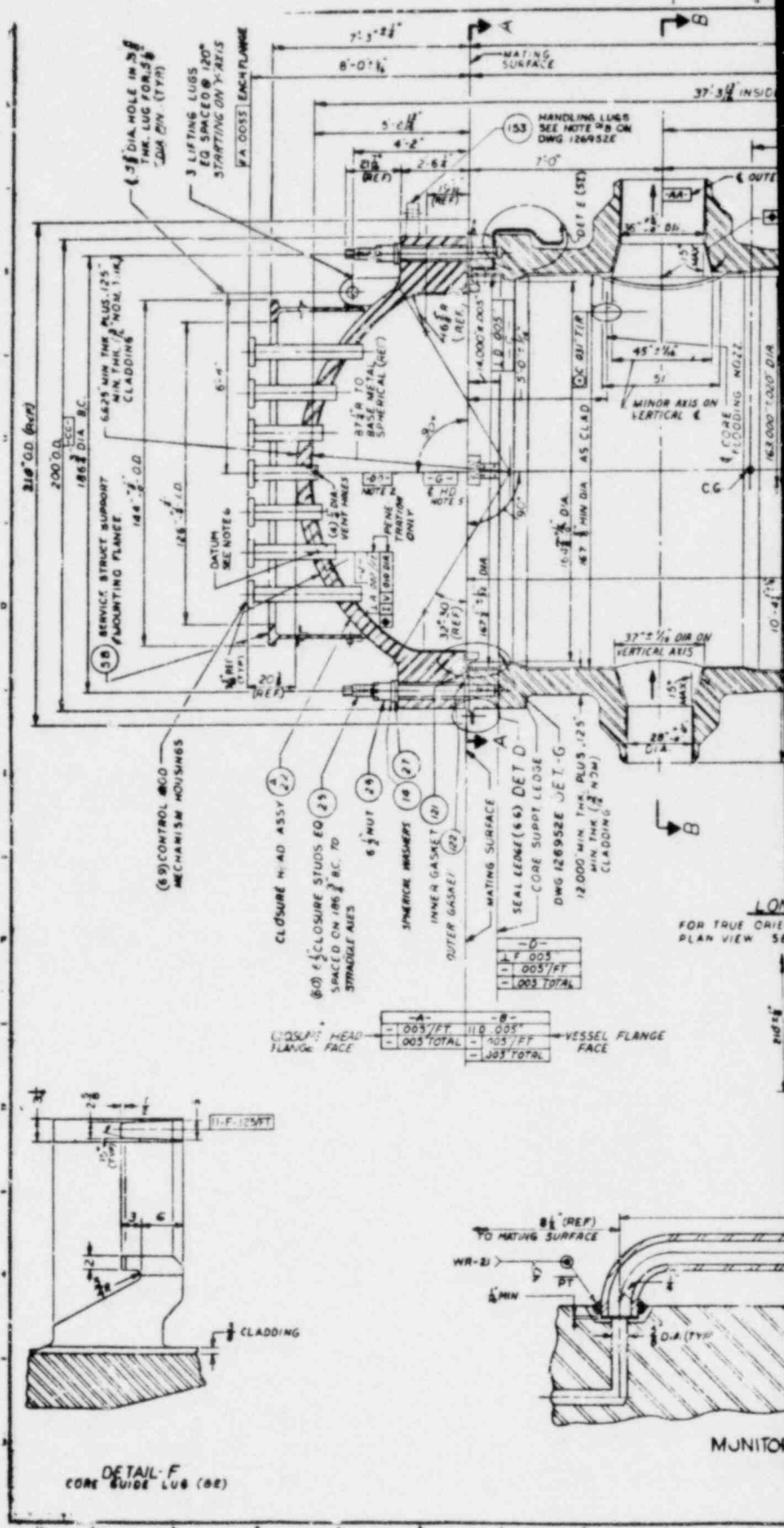
Shield plugs are situated to provide gaps capable of venting flow from the reactor vessel cavity to the service support structure cavity around the circumference of individual plugs. These small vent (gap) paths around the shield plugs were considered in the analysis. As such, form losses and frictional losses associated with flow over the sides of the plugs would be accounted for in the vent flow paths. Hence, drag-type losses due to flow along the height of a shield plug were accounted for. Thus, based on engineering judgement, the shield plug mass multiplier of 2 was introduced as a conservatism which delayed movement of the plugs (and correspondingly, reactor vessel cavity venting was delayed).

1.9. Supply drawings showing the reactor cavity and equipment and piping arrangement for the following plants: Crystal River 3 and Three Mile island Unit 1. (Section 4.3.2)

Response

Drawings are provided as follows:

	Figure No.	
	<u>Crystal River 3</u>	<u>TMI Unit 1</u>
Reactor vessel	1.9-1	1.9-5
Reactor coolant piping		
Elevation	1.9-2	1.9-6
Plan	1.9-3	1.9-7
RV primary shield wall	1.9-4	1.9-8



FOR TRUE ORIENT PLAN VIEW SEE...

MUNITOR





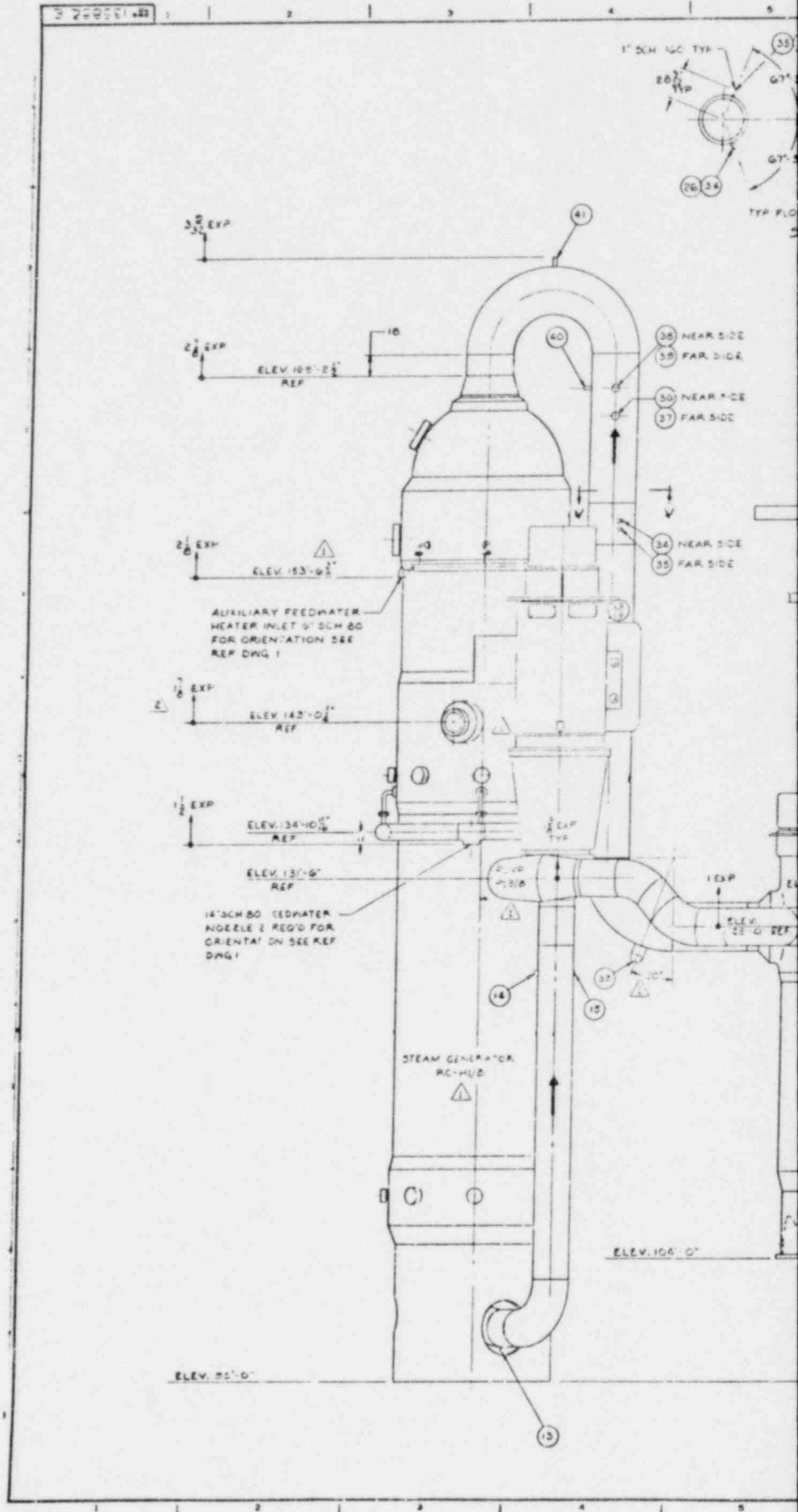
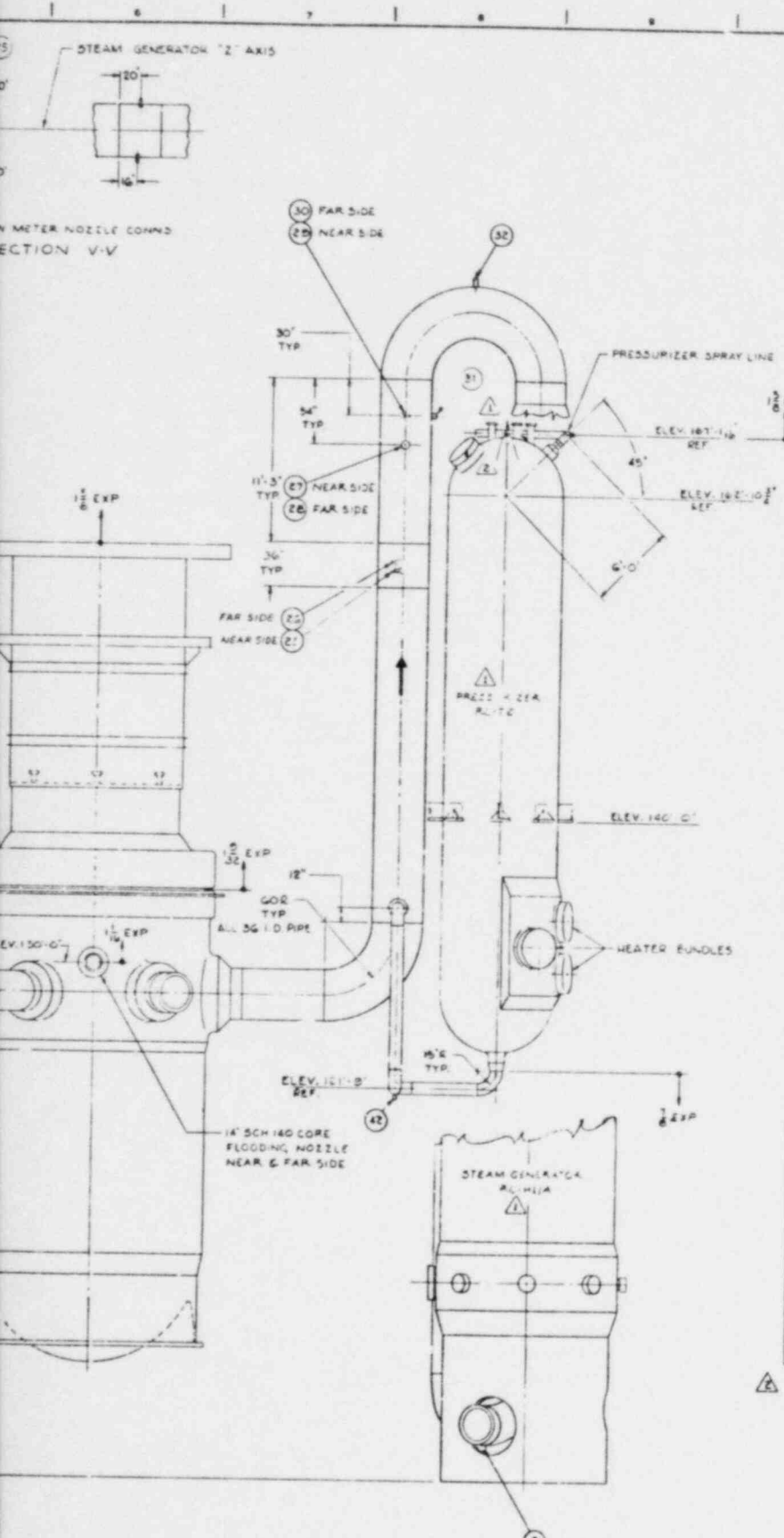


Figure 1.9-2



NO.	DESCRIPTION	DATE	APPROVED
1	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		
2	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		
3	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		
4	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		
5	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		
6	REVISED TO SHOW NOZZLE SIZES AND STANDARD RITE CONNECTIONS		

- NOTES
- FOR GENERAL NOTES AND PLAN VIEW SEE REF. DRG. 1
  - THERMAL EXPANSION INDICATED ON THIS DRAWING ARE FOR CLEARANCE PURPOSES ONLY AND NOT TO BE USED FOR ANY FLEXIBILITY ANALYSIS WORK. REFER TO SPECIFICATION 3007 WSS 7 FOR EXACT DISPLACEMENTS.

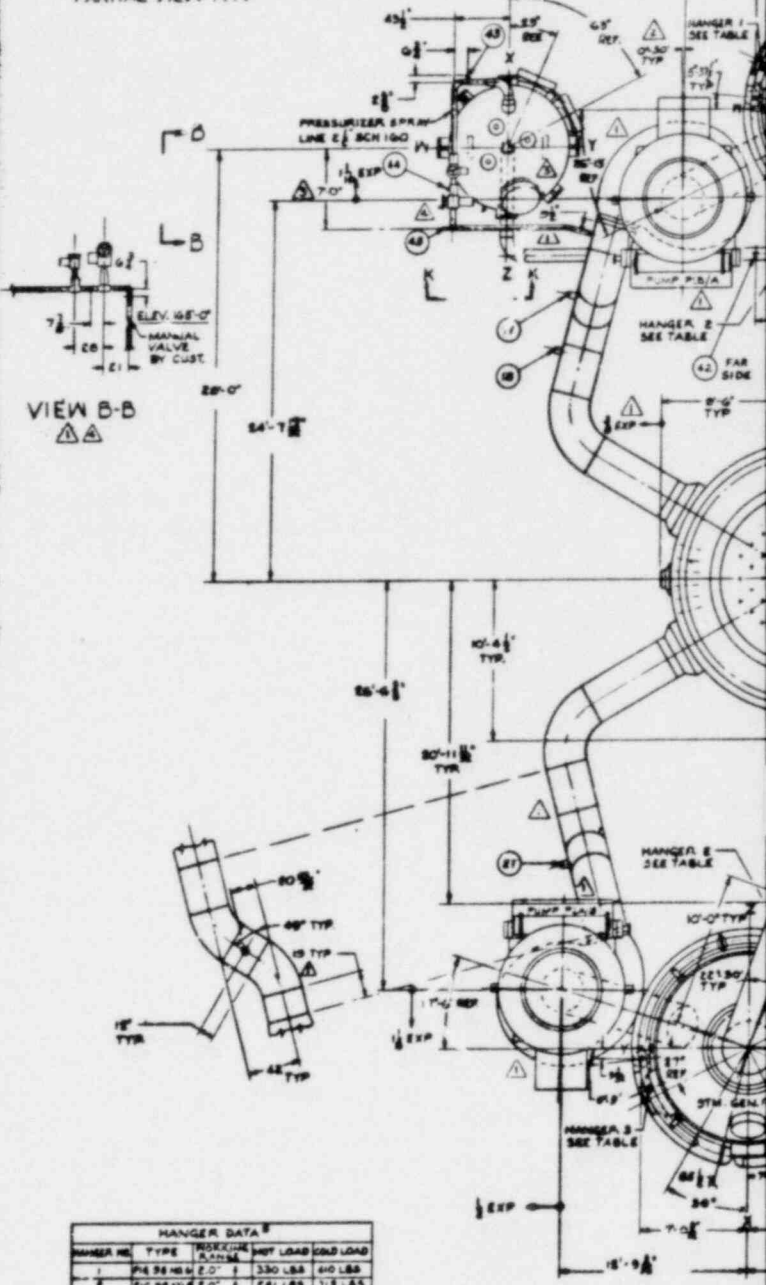
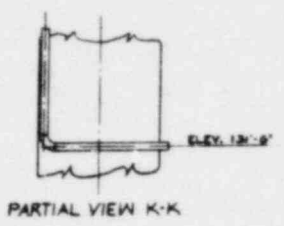
PIPING LEG	NOZZLE NO.	NOZ. PROJ. FROM ELEV. REF.	DESCRIPTION & SIZE	INSTANT NO.
STEAM GENERATOR RC HI A	1	20 1 4	DRAIN - 1" SCH 160	
TO	2	20 10 32	FAST RESPONSE RITE CONNECTION	103-50-104-50
TO	3	20	STANDARD RITE CONNECTION	102-50
PUMP P10 A	4	24 2 4	PRESS. TAP - 1" SCH 160	PI
STEAM GENERATOR RC HI B	5	20 1 4	DRAIN - 1" SCH 160	
TO	6	20 10 32	FAST RESPONSE RITE CONNECTION	101-50-102-50
TO	7	20	STANDARD RITE CONNECTION	101-50
PUMP P10 B	8	24 2 4	PRESS. TAP - 1" SCH 160	PI
STEAM GENERATOR RC HI C	9	20 1 4	DRAIN - 1" SCH 160	
TO	10	20 10 32	FAST RESPONSE RITE CONNECTION	103-50-104-50
TO	11	20	STANDARD RITE CONNECTION	102-50
STEAM GENERATOR RC HI D	12	20 1 4	DRAIN - 1" SCH 160	
TO	13	20 10 32	FAST RESPONSE RITE CONNECTION	101-50-102-50
TO	14	20	STANDARD RITE CONNECTION	101-50
PUMP P10 D	15	24 2 4	PRESS. TAP - 1" SCH 160	PI & LI
PUMP P10 E	16	20 1 16	HIGH PRESS. INJECTION - 2" SCH 160	
TO REACTOR VESSEL	17	24 2 4	PRESS. TAP - 1" SCH 160	PI & LI
PUMP P10 F	18	20 1 16	HIGH PRESS. INJECTION - 2" SCH 160	
TO REACTOR VESSEL	19	24 2 4	PRESS. TAP - 1" SCH 160	PI
PUMP P10 G	20	20 1 16	HIGH PRESS. INJECTION - 2" SCH 160	
TO REACTOR VESSEL	21	24 2 4	PRESS. TAP - 1" SCH 160	PI
PUMP P10 H	22	20 1 16	HIGH PRESS. INJECTION - 2" SCH 160	
TO REACTOR VESSEL	23	24 2 4	PRESS. TAP - 1" SCH 160	PI
REACTOR VESSEL	24	20 3 32	FLOWMETER CONN. - 1" SCH 160	PT1-100, PT1-101, PT1-102
TO	25	20 3 32	FLOWMETER CONN. - 1" SCH 160	PT1-103, PT1-104
STEAM GENERATOR RC HI A	26	24 10 32	FAST RESPONSE RITE CONNECTION	103-40-102-40
TO	27	24 10 32	FAST RESPONSE RITE CONNECTION	102-40-104-40
TO	28	24 10 32	FAST RESPONSE RITE CONNECTION	102-40-104-40
TO	29	24 10 32	FAST RESPONSE RITE CONNECTION	102-40-104-40
TO	30	20 1 4	STANDARD RITE CONNECTION	103-40-104-40
TO	31	20 2 8	STANDARD RITE CONNECTION	103-40
TO	32	20 3 8	STANDARD RITE CONNECTION	103-40
TO	33	20 3 8	STANDARD RITE CONNECTION	103-40
TO	34	20 3 8	STANDARD RITE CONNECTION	103-40
TO	35	20 3 8	STANDARD RITE CONNECTION	103-40
TO	36	20 3 8	STANDARD RITE CONNECTION	103-40
TO	37	20 3 8	STANDARD RITE CONNECTION	103-40
TO	38	20 3 8	STANDARD RITE CONNECTION	103-40
TO	39	20 3 8	STANDARD RITE CONNECTION	103-40
TO	40	20 3 8	STANDARD RITE CONNECTION	103-40
TO	41	20 3 8	STANDARD RITE CONNECTION	103-40
TO	42	10 7 8	DRAIN - 1" SCH 160	
TO	43	2 3 8	FOR SPRAY LINE - 1" SCH 160	
TO	44	5 1 4	BY-PASS CONN. - SCH 160	
TO	45	5 1 4	BY-PASS CONN. - SCH 160	

NOTE: THE PIPING TO THE LETDOWN COOLERS WILL HAVE 1/2" DIA. 1" TAPS DRAIN LINE

FLORIDA POWER CORPORATION  
CRYSTAL RIVER UNIT NO. 3

2 REACTOR COOLANT SYS P&ID SCHEMATIC 1 REACTOR COOLANT PIPING ARRANGEMENT NO. DRAWING TITLE REFERENCE DRAWINGS	35581 E DWG. NO. 600-0007-50 00	REACTOR COOLANT PIPING ARRANGEMENT ELEVATION CUSTOMER DRAWINGS	THE BAKER & WATSON COMPANY 135882 E 6
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108551



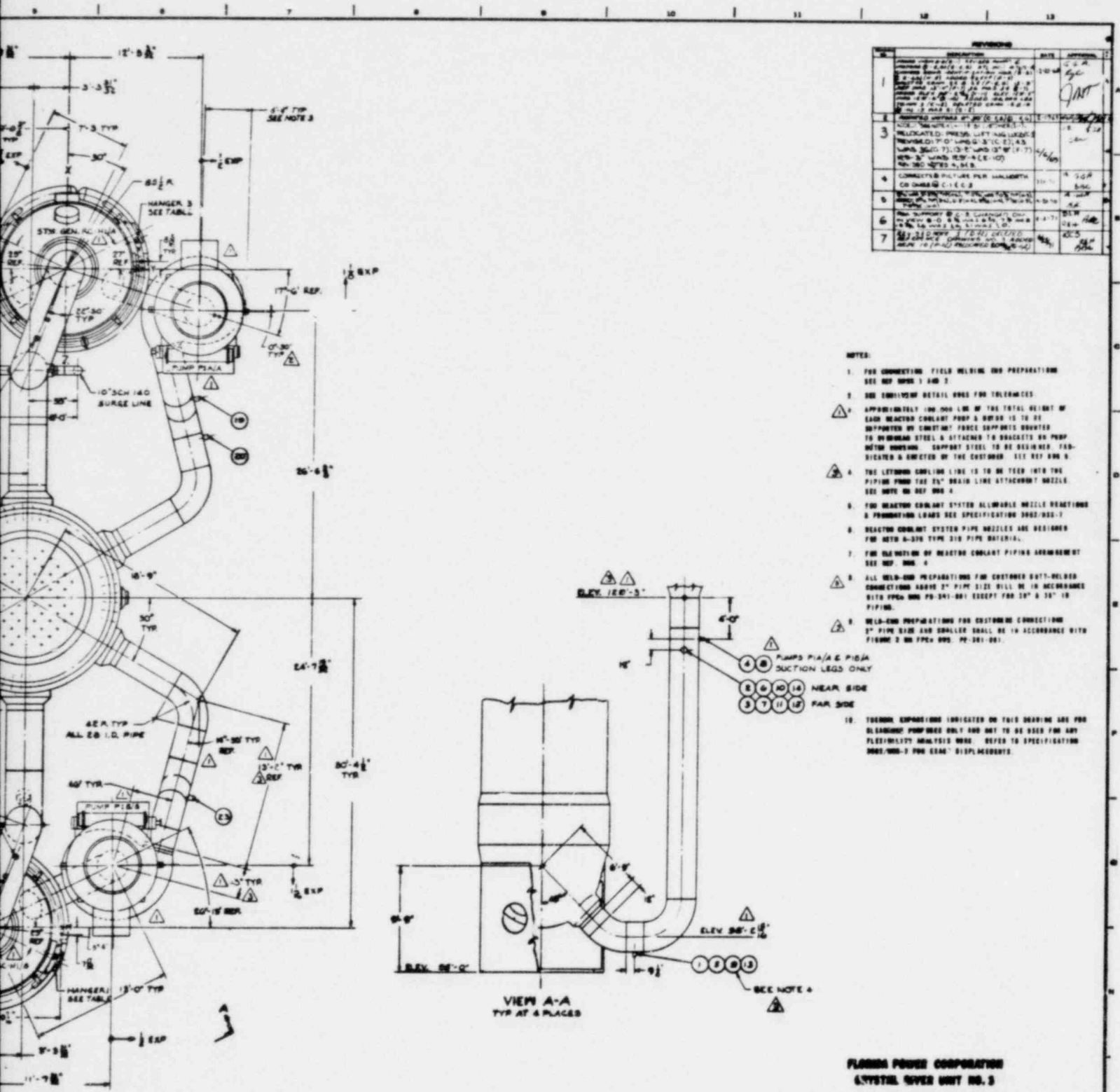
VIEW B-B

HANGER DATA

HANGER NO.	TYPE	TRUSS LINE	SPACING	NET LOAD	GROSS LOAD
1	P.1	20'-0"	2	330 LBS	410 LBS
2	P.2	20'-0"	2	281 LBS	352 LBS
3	P.3	20'-0"	2	288 LBS	359 LBS

\* CHANNEL C/S SPRING SUPPORT BY CUSTOMER

Figure 1.9-3



NO.	DESCRIPTION	DATE	APPROVAL
1	ISSUED FOR CONSTRUCTION	12-15-68	JNT
2	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
3	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
4	CONNECTED PICTURE FOR HALLWAY	1-15-69	JNT
5	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
6	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
7	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT

- NOTES:**
- FOR CONNECTIONS FIELD WELDING AND PREPARATIONS SEE REF DES 3 AND 2
  - SEE DIMENSIONAL DETAIL SHEETS FOR TOLERANCES
  - APPROXIMATELY 100,000 LBS OF THE TOTAL WEIGHT OF EACH REACTOR COOLANT PUMP IS TO BE SUPPORTED BY CONCRETE BRACE SUPPORTS MOUNTED TO OVERHEAD STEEL & ATTACHED TO BRACETS AS PROVIDED & ERECTED BY THE CUSTOMER. SEE REF DES 6.
  - THE LETHAL CHOKING LINE IS TO BE TYPED INTO THE PIPING FROM THE 2" BRAIN LINE ATTACHMENT NOZZLE. SEE NOTE ON REF DES 4.
  - FOR REACTOR COOLANT SYSTEM ALLOWABLE NOZZLE REACTIONS & FOUNDATION LOADS SEE SPECIFICATION 3002-003-1
  - REACTOR COOLANT SYSTEM PIPE NOZZLES ARE DESIGNED FOR ASTM A-334 TYPE 316 PIPE MATERIAL.
  - FOR SELECTION OF REACTOR COOLANT PIPING ARRANGEMENT SEE REF. DES. 4
  - ALL WELD-ON PREPARATIONS FOR CUSTOMER BUT-WELDED CONNECTIONS ABOVE 2" PIPE SIZE SHALL BE IN ACCORDANCE WITH SPEN AND PS-341-001 EXCEPT FOR 20" & 24" IN PIPING.
  - WELD-ON PREPARATIONS FOR CUSTOMER CONNECTIONS 2" PIPE SIZE AND SMALLER SHALL BE IN ACCORDANCE WITH FIGURE 2 ON SPEN AND PS-341-001.
  - TOEWORK EXPANSIONS INDICATED ON THIS DRAWING ARE FOR CLEARANCE PURPOSES ONLY AND NOT TO BE USED FOR ANY FLEXIBILITY ANALYSIS WORK. REFER TO SPECIFICATION 3002-003-1 FOR EXACT DIMENSIONS.

VIEW A-A  
TYP AT 4 PLACES

FLORIDA POWER CORPORATION  
CRYSTAL SPRING UNIT NO. 3

1	TYPE PUMP MOTOR SUPP. CRITERIA	048400	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
2	REACTOR COOLANT PIPING ARR. TELETYPE	15368 E	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
3	REACTOR COOLANT PIPING ASSY ELEVATION	14 146 B	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
4	REACTOR COOLANT PIPING ASSY PLAN	14 147 B	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
5	DRAWING TITLE	3002-003-00	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT
6	REFERENCE DRAWINGS	14 147 B	REVISED: PUMP PIA/A PIA/B SUCTION LEGS ONLY	1-15-69	JNT

See Appendix A for Figure 1.9-4, Drawing # SC-421-012

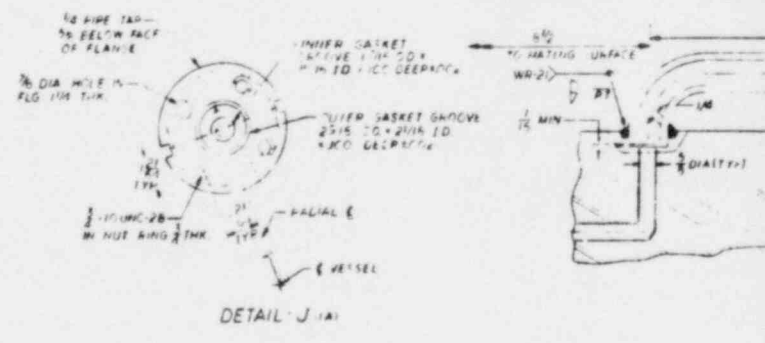
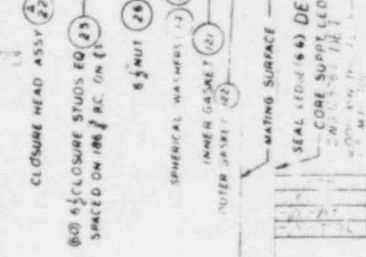
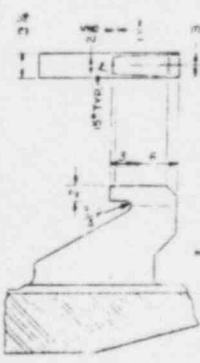
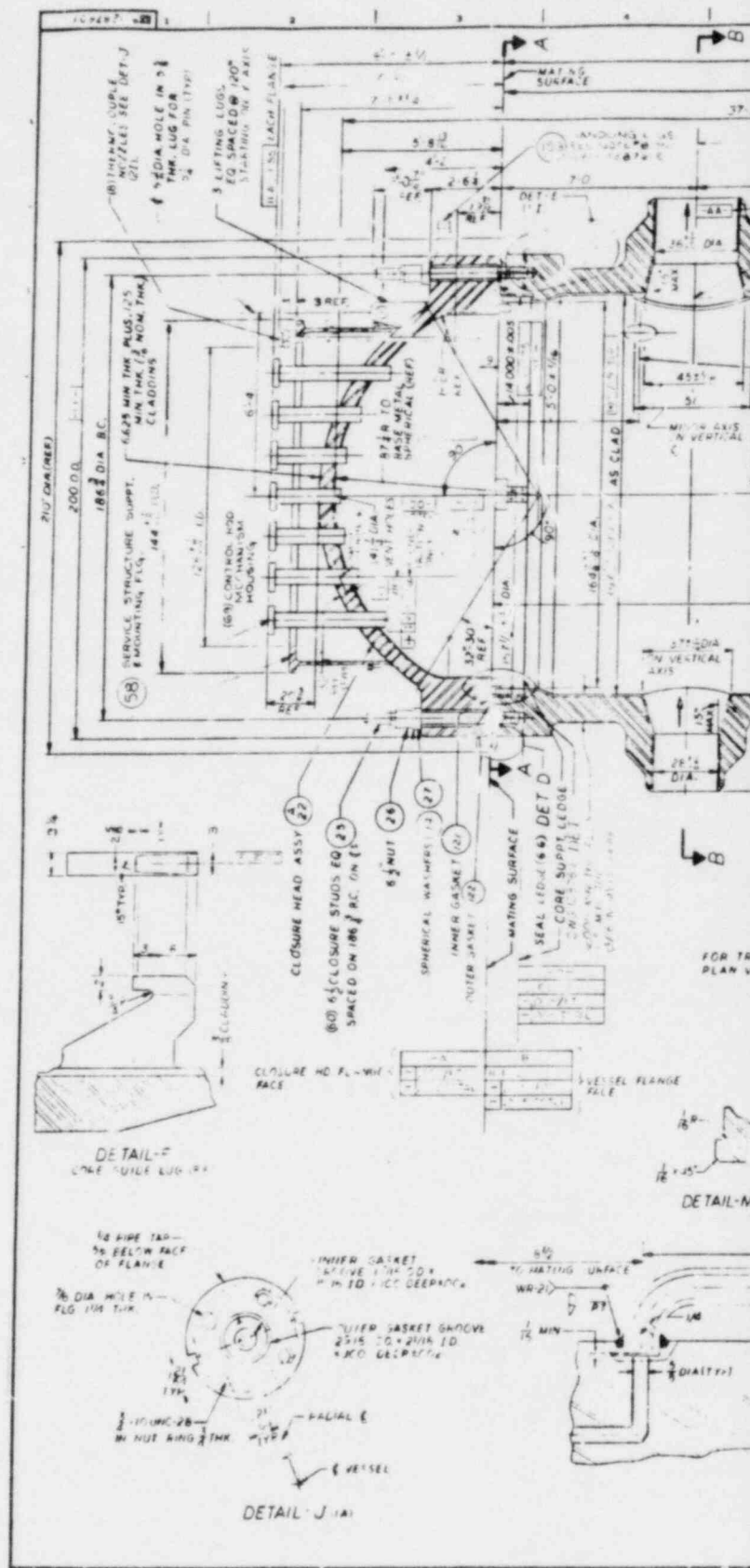
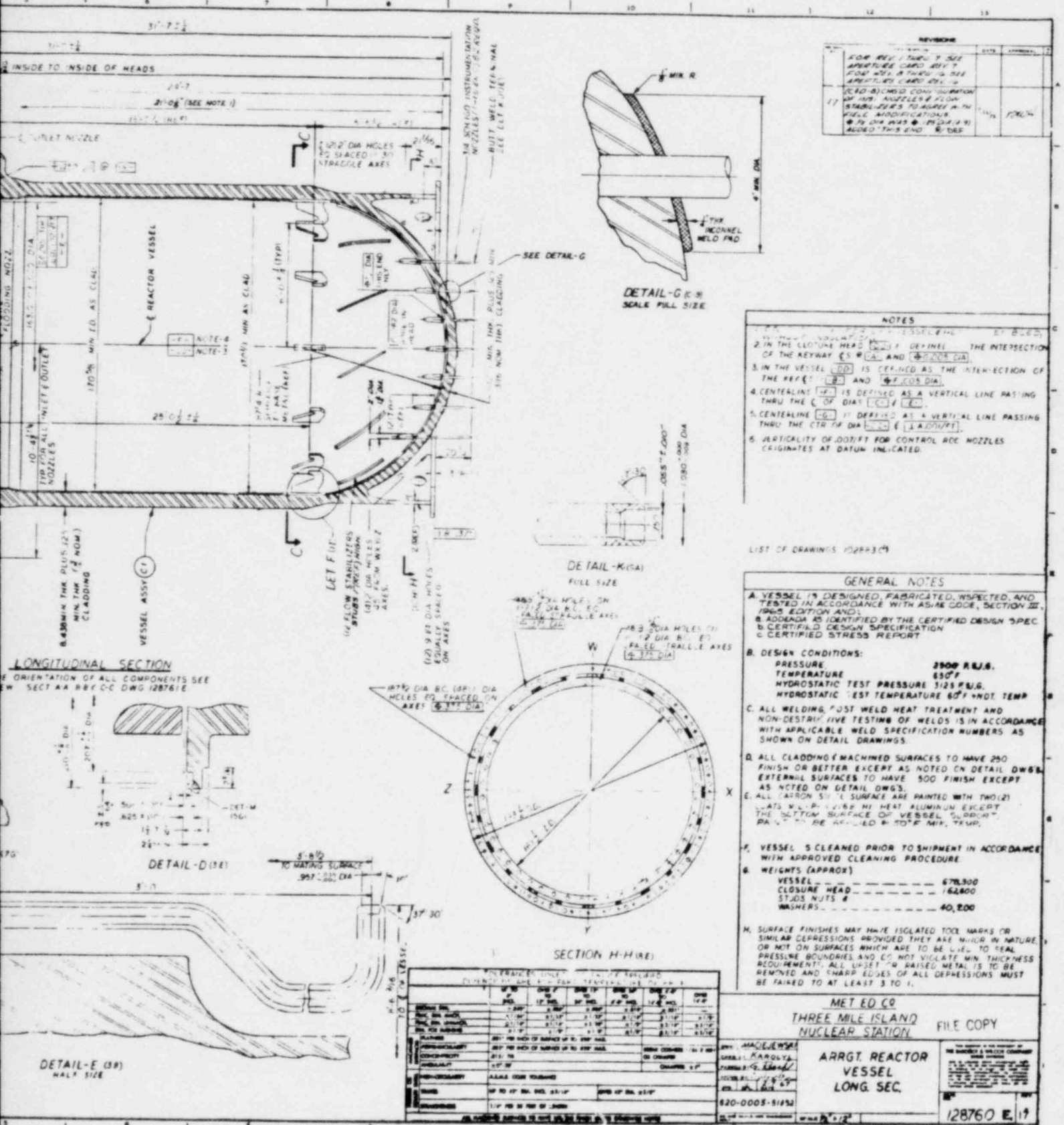


Figure 1.9-5



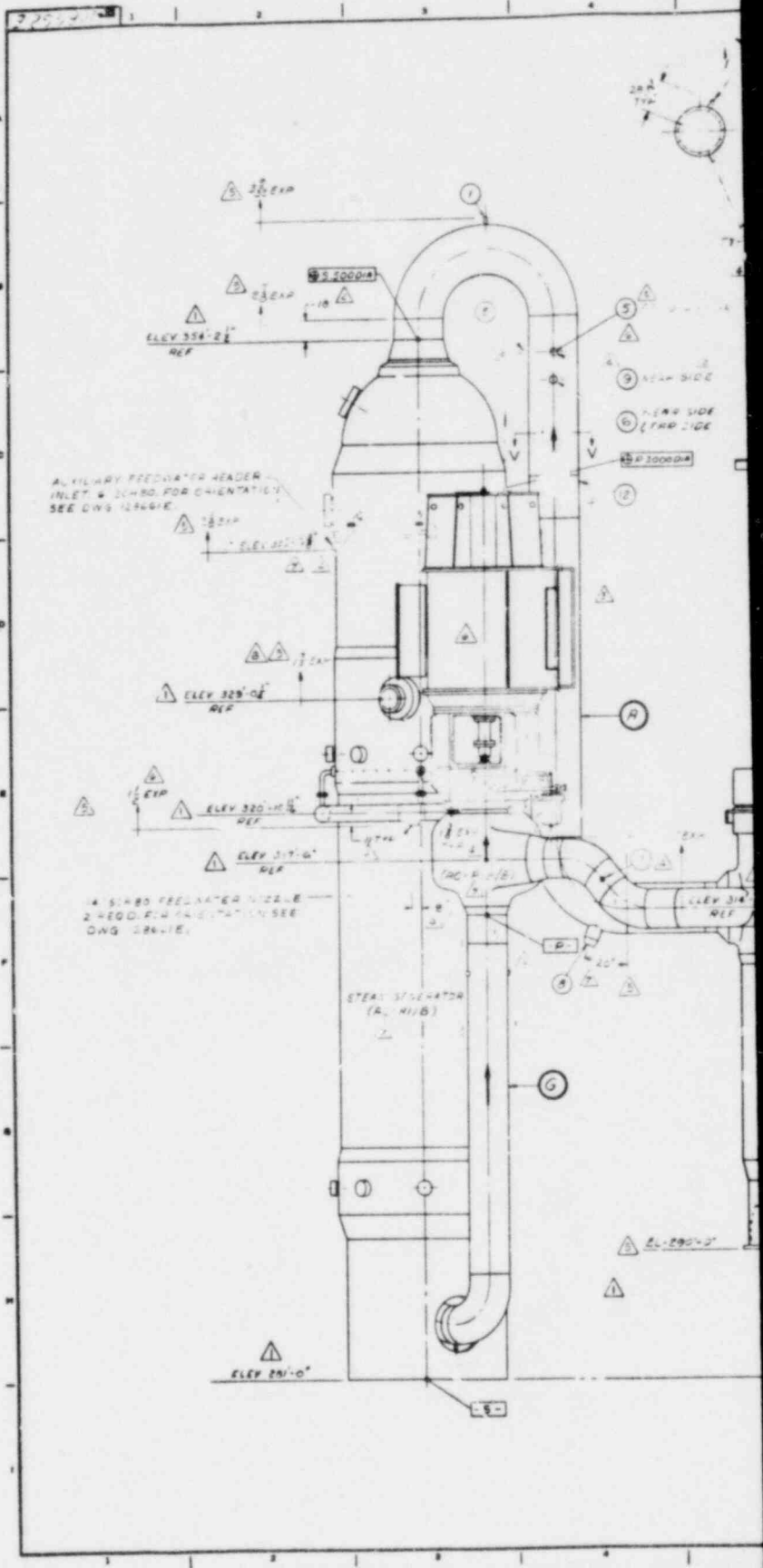
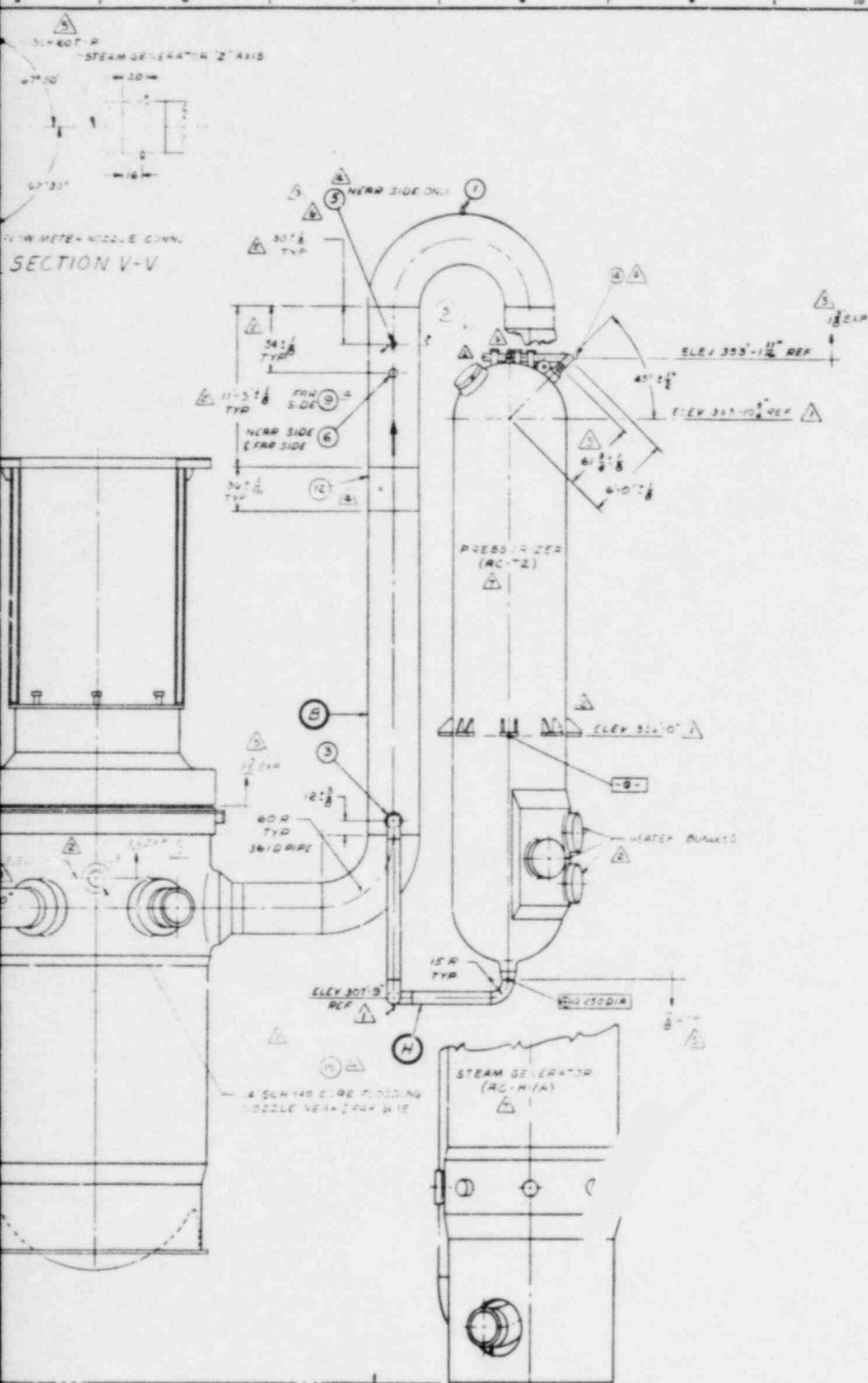




Figure 1.9-6



NO.	DESCRIPTION	DATE	APPROVAL
1	THIS DRAWING SUPERSEDES DWD NO. 1240'S & IS CHANGED FOLLOWING E.E. AT 01/10/68. DWD NO. 1240 WAS REVISED BY A. J. W. (12/13/67) ABOVE EDUCATIONAL REVISIONS. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
2	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
3	REVISED PUMP AND MOTOR DATA. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
4	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
5	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
6	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
7	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.
8	ADDED WITH SECTION V-V. (12/13/67) E.E. AT 01/10/68. ALL REVISIONS TO BE MADE BY THE DRAWING ENGINEER.	1/10/68	W. J. W.

NOTE:

1. FOR GENERAL NOTES, PLAN VIEW AND ASSEMBLY AND ATTACHMENT TABLE SEE DWG. 180661 E.

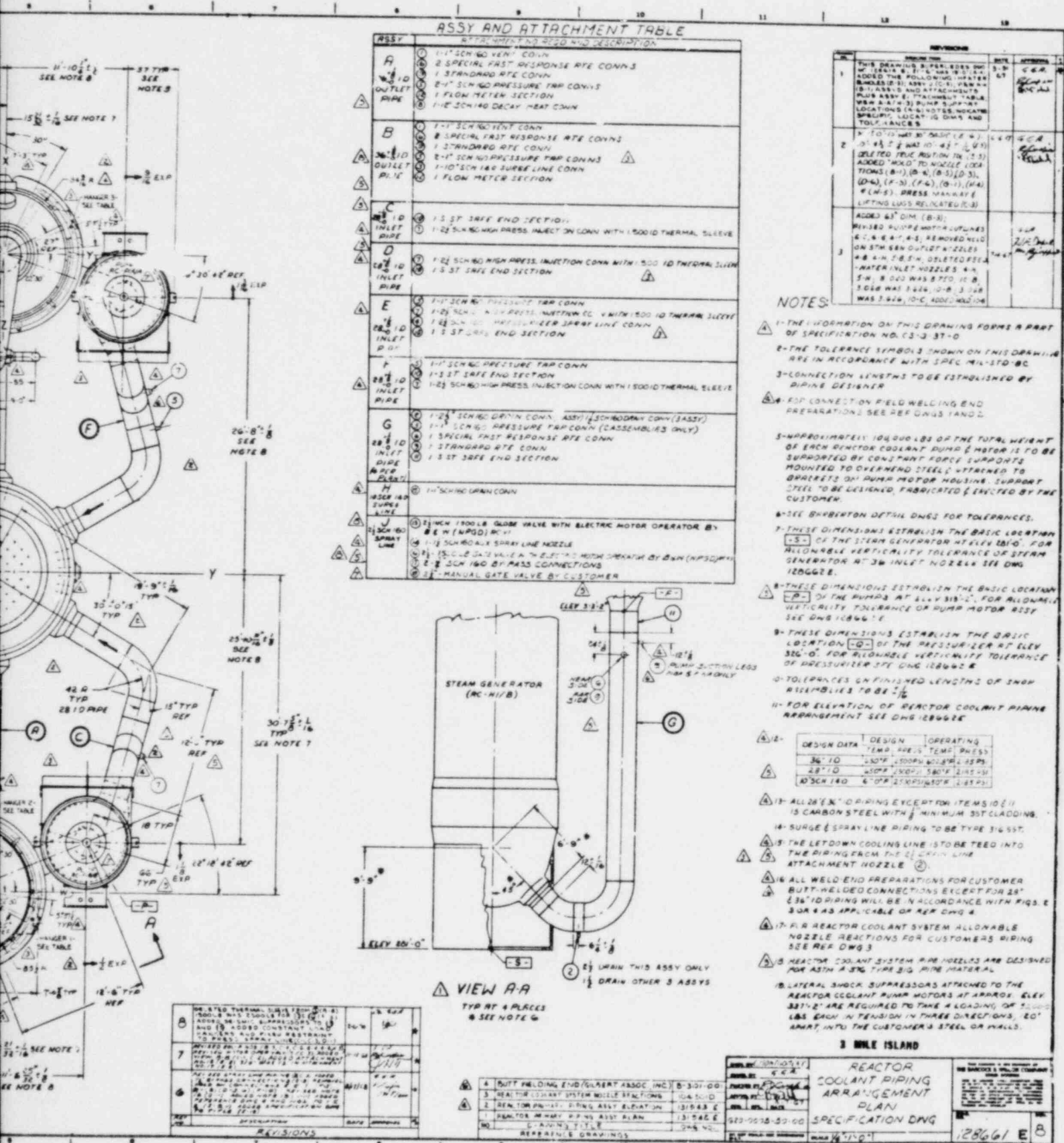
AS-BUILT DRAWING

3 MILE ISLAND

DESIGNED BY: W. J. W. CHECKED BY: W. J. W. DATE: 1/10/68	REACTOR COOLANT PIPING ARRANGEMENT ELEVATION SPECIFICATION DNG SCALE: 1/8" = 1'-0"	THIS DRAWING IS THE PROPERTY OF THE BARRON & WELCH COMPANY 1968
180661 E		180661 E



Figure 1.9-7



ASSY AND ATTACHMENT TABLE

ASSY	ATTACHMENT NO. REQ'D AND DESCRIPTION
A	1-1" SCH 80 VENT COVN 2 SPECIAL FAST RESPONSE RTE CONNS 1 STANDARD RTE COVN 2-1" SCH 80 PRESSURE TAP CONNS 1 FLOW METER SECTION 1-1" SCH 140 DECAT HEAT COVN
B	1-1" SCH 80 VENT COVN 2 SPECIAL FAST RESPONSE RTE CONNS 1 STANDARD RTE COVN 2-1" SCH 80 PRESSURE TAP CONNS 1-10" SCH 140 SURF LINE COVN 1 FLOW METER SECTION
C	1-3 ST SAFE END SECTION 1-22 SCH 80 HIGH PRESS INJECT ORN COVN WITH 1-500 ID THERMAL SLEEVE
D	1-22 SCH 80 HIGH PRESS INJECTION COVN WITH 1-500 ID THERMAL SLEEVE 1-3 ST SAFE END SECTION
E	1-1" SCH 80 PRESSURE TAP COVN 1-22 SCH 80 HIGH PRESS INJECT ORN COVN WITH 1-500 ID THERMAL SLEEVE 1-22 SCH 80 HIGH PRESS SURF LINE COVN 1-3 ST SAFE END SECTION
F	1-1" SCH 80 PRESSURE TAP COVN 1-3 ST SAFE END SECTION 1-22 SCH 80 HIGH PRESS INJECTION COVN WITH 1-500 ID THERMAL SLEEVE
G	1-1" SCH 80 VENT COVN (ASSY) / 1-22 SCH 80 VENT COVN (ASSTY) 1-1" SCH 80 PRESSURE TAP COVN (ASSEMBLY ONLY) 1 SPECIAL FAST RESPONSE RTE COVN 1 STANDARD RTE COVN 1-3 ST SAFE END SECTION
H	1-1" SCH 80 VENT COVN
I	1-22 SCH 140 GLOBE VALVE WITH ELECTRIC MOTOR OPERATOR BY 2 E W (NPOD) BY 1-1" SCH 80 SURF LINE NOZZLE 2-1" SCH 80 GLOBE VALVE WITH ELECTRIC MOTOR OPERATOR BY 2 E W (NPOD) BY 2-1" SCH 140 BY PASS CONNECTIONS 2-1" MANUAL GATE VALVE BY CUSTOMER

NO.	REVISIONS	DATE	BY	APPROVED
1	THIS DRAWING IS PUBLISHED BY THE U.S. GOVERNMENT AND IS NOT TO BE REPRODUCED IN ANY MANNER WITHOUT THE PERMISSION OF THE GOVERNMENT. THIS DRAWING IS THE PROPERTY OF THE U.S. GOVERNMENT AND IS NOT TO BE DISTRIBUTED OUTSIDE THE U.S. GOVERNMENT. THIS DRAWING IS THE PROPERTY OF THE U.S. GOVERNMENT AND IS NOT TO BE DISTRIBUTED OUTSIDE THE U.S. GOVERNMENT.	5-10-67	CCG	CCG
2	REVISED DIMENSIONS (A-1) TO (A-10) AND (B-1) TO (B-10) TO REFLECT THE ADDITION OF THE 'X' AND 'Y' COORDINATE SYSTEM. REVISED DIMENSIONS (C-1) TO (C-10) TO REFLECT THE ADDITION OF THE 'Z' COORDINATE SYSTEM. REVISED DIMENSIONS (D-1) TO (D-10) TO REFLECT THE ADDITION OF THE 'W' COORDINATE SYSTEM. REVISED DIMENSIONS (E-1) TO (E-10) TO REFLECT THE ADDITION OF THE 'V' COORDINATE SYSTEM. REVISED DIMENSIONS (F-1) TO (F-10) TO REFLECT THE ADDITION OF THE 'U' COORDINATE SYSTEM. REVISED DIMENSIONS (G-1) TO (G-10) TO REFLECT THE ADDITION OF THE 'T' COORDINATE SYSTEM. REVISED DIMENSIONS (H-1) TO (H-10) TO REFLECT THE ADDITION OF THE 'S' COORDINATE SYSTEM. REVISED DIMENSIONS (I-1) TO (I-10) TO REFLECT THE ADDITION OF THE 'R' COORDINATE SYSTEM.	6-10-67	CCG	CCG
3	REVISED DIMENSIONS (A-1) TO (A-10) TO REFLECT THE ADDITION OF THE 'X' AND 'Y' COORDINATE SYSTEM. REVISED DIMENSIONS (B-1) TO (B-10) TO REFLECT THE ADDITION OF THE 'Z' COORDINATE SYSTEM. REVISED DIMENSIONS (C-1) TO (C-10) TO REFLECT THE ADDITION OF THE 'W' COORDINATE SYSTEM. REVISED DIMENSIONS (D-1) TO (D-10) TO REFLECT THE ADDITION OF THE 'V' COORDINATE SYSTEM. REVISED DIMENSIONS (E-1) TO (E-10) TO REFLECT THE ADDITION OF THE 'U' COORDINATE SYSTEM. REVISED DIMENSIONS (F-1) TO (F-10) TO REFLECT THE ADDITION OF THE 'T' COORDINATE SYSTEM. REVISED DIMENSIONS (G-1) TO (G-10) TO REFLECT THE ADDITION OF THE 'S' COORDINATE SYSTEM. REVISED DIMENSIONS (H-1) TO (H-10) TO REFLECT THE ADDITION OF THE 'R' COORDINATE SYSTEM.	7-10-67	CCG	CCG

- NOTES:**
- 1-THE INFORMATION ON THIS DRAWING FORMS A PART OF SPECIFICATION NO. CS-3 37-0
  - 2-THE TOLERANCE SYMBOLS SHOWN ON THIS DRAWING ARE IN ACCORDANCE WITH SPEC MIL-STD-88C
  - 3-CONNECTION LENGTHS TO BE ESTABLISHED BY PIPING DESIGNER
  - 4-EXP. CONNECTION FIELD WELDING END PREPARATIONS SEE REF DWG 1 AND 2
  - 5-WPDR APPROXIMATELY 100,000 LBS OF THE TOTAL WEIGHT OF EACH REACTOR COOLANT PUMP MOTOR IS TO BE SUPPORTED BY COND. PART HOIST SUPPORTS MOUNTED TO OVERHEAD STEEL STRUCTURE TO BRACKETS ON PUMP MOTOR HOUSING. SUPPORT STEEL TO BE DESIGNED, FABRICATED & ERECTED BY THE CUSTOMER.
  - 6-SEE BRIBERTON DETAIL DWGS FOR TOLERANCES.
  - 7-THESE DIMENSIONS ESTABLISH THE BASIC LOCATION OF THE STEAM GENERATOR AT ELEV 120'-0" FOR ALLOWABLE VERTICALITY TOLERANCE OF STEAM GENERATOR AT 36" INLET NOZZLE SEE DWG 128662 E
  - 8-THESE DIMENSIONS ESTABLISH THE BASIC LOCATION OF THE PUMPS AT ELEV 318'-2" FOR ALLOWABLE VERTICALITY TOLERANCE OF PUMP MOTOR ASSEMBLY SEE DWG 128662 E
  - 9-THESE DIMENSIONS ESTABLISH THE BASIC LOCATION OF THE PRESSURIZER AT ELEV 352'-0" FOR ALLOWABLE VERTICALITY TOLERANCE OF PRESSURIZER SEE DWG 128662 E
  - 10-TOLERANCES ON FINISHED LENGTHS OF INSP ASSEMBLIES TO BE ± 1/16"
  - 11-FOR ELEVATION OF REACTOR COOLANT PIPING ARRANGEMENT SEE DWG 128662 E
  - 12-DESIGN DATA: DESIGN TEMP. 445.5° F, PRESS. 36.10 LBS/IN<sup>2</sup>, COOLANT FLOW RATE 1200 GPM, SURF. AREA 140.0 SQ. FT., SURF. AREA 140.0 SQ. FT.
  - 13-ALL 24" SCH 40 PIPING EXCEPT FOR ITEMS I & J IS CARBON STEEL WITH 1/8" MINIMUM SST CLADDING.
  - 14-SURGE & SPRAY LINE PIPING TO BE TYPE 316 SST.
  - 15-THE LETDOWN COOLING LINE IS TO BE TIED INTO THE PIPING FROM THE 24" CRAN LINE ATTACHMENT NOZZLE
  - 16-ALL WELD END PREPARATIONS FOR CUSTOMER BUTT-WELDED CONNECTIONS EXCEPT FOR 24" & 36" ID PIPING WILL BE IN ACCORDANCE WITH FIGS. 2 & 3 OR AS APPLICABLE OF REF DWG 4
  - 17-FOR REACTOR COOLANT SYSTEM ALLOWABLE NOZZLE REACTIONS FOR CUSTOMER'S PIPING SEE REF DWG 3
  - 18-REACTOR COOLANT SYSTEM PIPING NOZZLES ARE DESIGNED FOR ASTM A 312 TYPE 316 PIPE MATERIAL
  - 19-LATERAL SHOCK SUPPRESSORS ATTACHED TO THE REACTOR COOLANT PUMP MOTORS AT APPROX. ELEV. 327'-2" ARE REQUIRED TO TAKE A LOADING OF 5000 LBS EACH IN TENSION IN THREE DIRECTIONS, 20° APART, INTO THE CUSTOMER'S STEEL OR WALLS.

VIEW A-A  
TYP AT 4 PLACES  
SEE NOTE 4

NO.	DESCRIPTION	DATE	BY	APP'D
1	REVISED DIMENSIONS (A-1) TO (A-10) AND (B-1) TO (B-10) TO REFLECT THE ADDITION OF THE 'X' AND 'Y' COORDINATE SYSTEM.	5-10-67	CCG	CCG
2	REVISED DIMENSIONS (C-1) TO (C-10) TO REFLECT THE ADDITION OF THE 'Z' COORDINATE SYSTEM.	6-10-67	CCG	CCG
3	REVISED DIMENSIONS (D-1) TO (D-10) TO REFLECT THE ADDITION OF THE 'W' COORDINATE SYSTEM.	7-10-67	CCG	CCG

DESIGNED BY: CCG	CHECKED BY: CCG	DATE: 5-10-67
DRAWN BY: CCG	DATE: 5-10-67	SCALE: 1/8" = 1'-0"
3 MILE ISLAND REACTOR COOLANT PIPING ARRANGEMENT PLAN SPECIFICATION DWG 128661 E		

See Appendix A for Figure 1.9-8, Drawing # E-421-016

1.10. Provide an assessment of the influence of vena contracta effects on the reported results. (Section 4.3.2)

Response

As part of the A-2 program, B&W participated in numerous meetings with the NRC. One topic considered in great detail was the methodology to be used in generating asymmetric reactor cavity pressures. The result of these discussions were that B&W would use the CRAFT2 computer code<sup>1</sup>, that all flow paths would be checked for critical flow using the Moody correlation with  $C_D = 0.6$ , and that minimum geometric flow path areas would be employed.

Consider also that most area changes within the reactor vessel cavity are gradual in nature, so that vena contracta effects should not be significant. Furthermore, the modeling methodology incorporates demonstrated flow path area and loss coefficient conservatisms.

In view of the discussion above, vena contracta effects would not significantly affect the results.

1.11. What is the source document for the ACP and jet impingement loading data? (Section 11.1.3)

This question was revised at the April 2-3 meeting to request a tabulation of peak ACF forces from the available data.

Response

Table 11.1-1 of BAW-1621 provides the forces before multiplication by any factors, and an example of JI loads is given in Table 11.1-2.<sup>2</sup> The attached Table 1.11-1 provides the requested peak forces and moments. When the cold leg is broken in this table, it is in the same leg as the P1A1 pump. Also, all breaks are 1.0A except the upper hot leg breaks for the Midland plant, which are 2.0A.

Table 1.11-1. Peak Forces and Moments

Component	Peak forces/moments, -kips/ft-kips											
	Break @ pump discharge			Break @ pump suction			Break @ lower hot leg			Break @ upper hot leg		
	FX	FY	FZ	FX	FY	FZ	FX	FY	FZ	FX	FY	FZ
<u>Midland 1</u>												
PIA1 casing	-120.6 <sup>a</sup>	201.5 <sup>b</sup>	-167.3 <sup>a</sup>	28.4	150.2	24.4				-5.0	114.8	11.1
MS	-6.6	N	11.7	-6.2	N	6.6		NA		18.0	N	23.6
M	-11.1	N	19.7	-10.5	N	11.1				30.4	N	39.8
MT	-4.5	N	7.9	-4.2	N	4.5				12.2	N	16.0
PIA2		NA			NA			NA			NA	
OTSG LO	309.7 <sup>c</sup>	N	-206.9 <sup>c</sup>	292.3	N	-49.7				26.0	N	-18.9
MID	140.6	N	-145.7	347.1	N	-90.8		NA		-35.8	N	34.2
UP	94.7	N	-113.6	128.0	N	-4.0				160.0	N	-99.0
UH	9.8	N	4.7	9.9	N	2.7				53.4	N	-262.5
<u>DB-2, 3</u>												
PIA1 F	301.0 <sup>d</sup>	329.4 <sup>e</sup>	601.2 <sup>d</sup>	-74.7	445.0	129.2	-171.5	274.0	208.5			
M	6779.0	N	3395.0	1904.0	N	-494.0	2446.0	N	1709.0		NA	
PIA2 F	113.0	26.7	30.9		NA		266.9	257.1	117.8		NA	
M	533.7	NA	1792.0				1280.0	N	3037.0			
OTSG		NA			NA		153.6 <sup>f</sup>	N	808.5 <sup>f</sup>	19.4	N	246.3
							9599.0	N	1831.0	12220.0	N	1062.0

<sup>a</sup>Figure 11.1-9.

<sup>b</sup>Figure 11.1-10 – figure and peak do not agree because peak is net force, not just upward.

<sup>c</sup>Figure 11.1-11.

<sup>d</sup>Figure 11.1-12.

<sup>e</sup>Figure 11.1-13.

<sup>f</sup>Figure 11.1-14 – error in graph; should be  $\times 10^6$ .

Legend – NA: not available (same as Table 11.1-1), N: negligible or not applicable, MS: motor stand, M: motor, MT: top of motor, LO: discharge lower head, MID: midsection of OTSG, UP: upper tubesheet, UH: upper head.

- 1.12. Assess asymmetric cavity pressure loadings for which ACP data were not available, including breaks at the steam generator outlet nozzle. (Section 11.1-3)

Response

Terminal end guillotine breaks were considered at the steam generator inlet and outlet nozzles and at the PIA2 pump suction and discharge nozzles (see Table 1.12-3). Data from available ACP cases were applied to locations where data were not available, e.g., pump suction data were applied at the OTSG outlet. Tables 1.12-1 and 1.12-2 show which data from Table 1.12-3, with appropriate multiplication factors, were considered at all locations for both nozzle- and skirt-supported plants. Davis-Besse 1 does not require the 1.5 factor for steam generator compartment configuration applied in Section 11.1-3<sup>2</sup> since Davis-Besse 2 and 3 would have been identical.

At the April 2-3 meeting with NRC and EG&G, the specific emphasis on this question was consideration of a break at the OTSG discharge nozzle. Based on the following consideration, a pressure buildup inside the OTSG support skirt would have essentially no impact on OTSG loadings and thus would not be the design loading case.

Due to the nozzle/pipe interface location in the 177-FA plants with respect to the OTSG support skirt, discharge into the OTSG support skirt cavity would be minimal for a 2A break (see Figure 1.9-3, view A-A). The steam generator side of the break would discharge out through the piping penetration of the support skirt since the OTSG/support skirt relationship would remain fixed during a LOCA. The pump suction side of the break must physically move away from the OTSG nozzle to create a 2A BOA, and this movement would cause most or all of the discharge from the break to be outside the OTSG support skirt. Therefore, the break inside the OTSG support was judged to be bounded by the breaks analyzed.



Table 1.12-1. ACP Data for Lowered-Loop/Skirt-Supported Plants

	<u>Pump disch</u>	<u>Pump suction</u>	<u>OTSG inlet</u>	<u>OTSG outlet</u>
PIA1 broken piping	3A	3C	1.5E	3C
PIA2 unbroken piping	3A	3C	1.5E	3C
OTSG	3B	3D	1.5F	3D

Note: Cases A-F are from Table 1.12-3.

Table 1.12-2. ACP Data for Raised-Loop/Nozzle-Supported Plant

	<u>Pump disch</u>	<u>Pump suction</u>	<u>OTSG inlet</u>	<u>OTSG outlet</u>
PIA1 broken pump	2G	2I	2J	2I
PIA2 unbroken pump	2H	2I	2K	2I
OTSG	2L	2L	2M	2L

Note: Cases G-M are from Table 1.12-3.

Table 1.12-3. Available ACP Cases From Table 11.1-1<sup>2</sup>

<u>Component</u>	<u>Break @ pump disch</u>	<u>Break @ pump suction</u>	<u>Break @ lower hot leg</u>	<u>Break @ upper hot leg</u>
<u>Midland 1</u>				
P1A1 pump	1A (A)	1A (C)	na	2A (E)
P1A2 pump	na	na	na	na
OTSG	1A (B)	1A (D)	na	2A (F)
<u>Davis-Besse 2, 3</u>				
P1A1 pump	1A (G)	1A (I)	1A (J)	na
P1A2 pump	1A (H)	na	1A (K)	na
OTSG	na	na	1A (L)	1A (M)

- Notes: 1. A = cross-sectional flow area of pipe.  
 1A = 642.5 in.<sup>2</sup> for pump suction or discharge break.  
 1A = 1018 in.<sup>2</sup> for upper or lower hot leg break.
2. na = not available.

- 2.1. In Figures 8.2-1, 8.2-4, 8.2-7, and 8.2-10, identify the interface across which the RV head differential pressures are determined (refer back to Appendix C Figures C-2 and C-9)

Response

Reactor vessel internal head differential pressure is the pressure difference between the lower and upper heads. Figures 8.2-1 and 8.2-4 are for the skirt-supported plants, and the differential pressure is between nodes 141 and 157 shown in Figure C-2 of Appendix C.<sup>2</sup> Figures 8.2-7 and 8.2-10 are for the nozzle-supported plant, and the differential pressure is between nodes 5 and 10, shown in Figure C-9 of Appendix C.<sup>2</sup>

- 2.2. Identify and discuss out-of-RV cavity pipe break locations considered or evaluated for RV internals hydraulic transients. Apparently in section 8.2, RV internals loads were calculated for the SG cavity break — what was the reason for not identifying this condition in section 4.4? Confirm that out-of-RV cavity BOAs in Section 11 are not limiting relative to RV internals hydraulic transients.

Response

In section 8.2, two hot leg and two cold leg breaks were considered. The hot and cold leg breaks at the RV nozzle are in the RV cavity. The other two breaks, at the entrance to the lower hot leg elbow and at the cold leg elbow exit, as illustrated in Figure 4.1-1, are located in the steam generator cavity. These breaks in the steam generator cavity are identified in section 4.4, first paragraph, last sentence, as "the next closest (to the reactor vessel) hot leg and cold leg break locations."

For consistent BOA/BOT and break type, breaks at the reactor vessel would be expected to be limiting on the RV internals since neither jet impingement nor cavity pressures is present for breaks outside the RV cavity. Also, pipe whip restraints in the steam generator compartment should reduce BOA, but in the interest of conservatism, they were not considered, and evaluations in section 11 were for 2A BOAs only.<sup>2</sup>

2.3. Were high-pressure injection ECCS piping and core flood line check valves considered or analyzed for hydraulic transients? Discuss load severity relative to RCS break size and location.

Response

The structural evaluation of the unbroken primary piping demonstrated that the integrity of the piping was maintained for the postulated breaks within the reactor subcompartment. The high-pressure injection piping is attached directly to the primary piping and was therefore considered acceptable since no large displacements were imparted to it by the primary piping.

The core flood line analysis considered reactor vessel displacement motion due to asymmetric LOCA loadings and was found acceptable. Table 9.9-1 of reference 2 gives the maximum acceleration and maximum displacement at the core flood nozzle for all plants. It can be seen that, based on resulting core flood nozzle stresses, the hot leg break at the reactor vessel on TMI-1, CR-3 and Davis Besse 1 is controlling.

2.4. Discuss the core flood line load severity relative to RCS break location and size.

Response

The core flood line load severity relative to a cold leg guillotine break at the RV nozzle on the RV internals is very small. A 1A cold leg break is 615.75 inches square, whereas a 1A core flood line nozzle break is only 107.51 inches square, or a factor of 5.7 smaller than the 1A cold leg break. A 2A core flood line break would be equivalent in size to a 0.35A cold leg break but would discharge less energy because the core flood tank side of the break is at a lower temperature and pressure than the primary side. Also a core flood line break would only discharge reactor coolant from one side of the break, thus having even less effect on RV internals than a hot or cold leg break.

Also see the response to question 2.3 concerning controlling cases.

2.5. In Table 8.2-2, discuss why the RV head differential pressure is higher for a 1.5A cold leg break rather than for a 2.0A cold leg break. Would a corresponding relationship (as shown) exist for a 1.5A SG cavity cold leg break?

Response

The results reported in Table 8.2-2 of reference 2 for the RV head differential pressure are attributable to the vent valves. Vent valves provide communication between the hot interior of the RV and its relatively cold down-comer region. As such, the valves are continually responding to fluid changes and quickly propagate any such changes to the RV upper and lower heads. This makes the RV head differential pressure more susceptible to large amplitude oscillations.

Since the RV head differential pressure is clearly designed by hot leg breaks (see Table 8.2-1 in reference 2, "Reactor Head Pressure Differential" data), and since breaks in the steam generator cavity exert no asymmetric loadings on the RV shell, variations in the RV head differential pressures for cold leg breaks in the RV or the steam generator cavities are enveloped by other cases.

2.6. It is not clear which BOAs are used for each specific analysis for the skirt- and nozzle-supported plants. In a table, summarize the break location, size, time, and type for each RV internals, ECCS, RCS, and cavity analysis (refer to Figures 4.2-3 through 4.2-17, as applicable). See questions 7, 8, 9, and 10.

#### Response

The derivation of generic break opening area (BOA) versus break opening time (BOT) curves and their use in the analysis of the B&W 177-FA Owners Group plants is described below.

Generic BOA versus BOT curves are to be used to generate hydraulic forcing functions in a conservative manner without making ultra-conservative assumptions for the BOTs for specific BOAs. These curves also allowed the hydraulic forcing functions to be generated before actual plant specific BOA versus BOT curves were available. Thus, a catalog of BOA hydraulic data was generated for specific BOAs (i.e., 0.3A, 0.6A, 1.0A, 1.5A, 2.0A), and the generic curves were used to determine the BOT rate up to each of the BOAs in the catalog.

The generic curves are developed to be conservative yet give realistic BOTs. As a general rule the faster the break opens, the worse the asymmetric pressures on the internals. Also, the total integrated mass and energy are greater for the break that opens more quickly. The only place a faster opening rate is not a worse case is on the broken pipe. The force on the broken pipe decays as a function of the system pressure times the inside cross-sectional area of the pipe ( $P \cdot A$ ), and the faster the break opens the faster this decay in the first elbow away from the break (the force generated in the first elbow controls the major portion of the pipe whip). Thus, in a systematic way, BOA versus BOT curves were generated which opened at a realistic and conservatively rapid rate for use in calculating internal asymmetric pressures and mass and energy.

The following is a description of the nonlinear pipe whip analysis used to generate these conservative BOA versus BOT curves:

There were two basic piping configurations for the plants analyzed. The first configuration applies to the skirt-supported plants. The hot and cold leg piping dimensions are generally the same for these plants, with only minor differences. These minor differences do not impact the generic BOA versus BOT



curves since they depend on the overall flexibility of the piping from the RV to the steam generator. The pipe whip restraint configurations on each plant are also different. However, they were considered negligible in generating the generic curves.

The other basic loop piping configuration is that of the nozzle-supported plant. Again, the pipe whip restraints were considered to have negligible impact in generating the generic curves.

The following model types were used to obtain the generic area versus time curves.

1. Hot leg piping geometry for skirt-supported plant (Figures B-1 through 6 in reference 2).
2. Cold leg piping geometry for skirt-supported plant (Figure B-9 in reference 2).
3. Hot leg piping geometry for nozzle-supported plant (Figure B-7 in reference 2).
4. Cold leg piping geometry for nozzle-supported plant (Figure B-8 in reference 2).

The models used in the ANSYS pipe whip analysis to obtain the generic curves included no restraints to impede pipe motion. The piping runs were modeled from the reactor vessel to the steam generator and the cold leg, including the reactor coolant pump. The reactor vessel and steam generator contribution to the motion of the whipping pipe were considered negligible due to mass of the steam generator and the time domain of interest for the pipe whip and thus were fixed dynamically or left out. Thermal and deadweight displacements for the reactor vessel and the OTSG were obtained from previous analyses of plants with the same configurations and were applied as displacements at the boundary joints. The thermal and deadweight, along with the pressure stresses in the piping runs, were addressed via thermal expansion, weight effects, and system pressure being applied to the piping run in concert with the boundary displacements.

Investigations were conducted as to which break locations resulted in the fastest opening rates for the unrestrained pipe whip. For the break locations addressed in section 4.1 of reference 2 (e.g., hot leg RV nozzle, hot leg lower elbow entrance, cold leg RV nozzle, and cold leg elbow exit closest to the RV),

the breaks at the reactor vessel nozzles were found to result in the quickest opening rates. Thus, breaks at the RV nozzles for the hot and cold legs were run for both plant configurations.

For the breaks considered, the major forcing function on the pipe is in the first elbow and basically results in a force at time  $0^+$  (i.e., after instantaneous pipe severance) of the operating pressure times the flow area of the pipe ( $P \cdot A$ ). This force decays as the pressure in the elbow decreases (i.e., as the break opens). In order to predict the most rapid opening rates, force decay was neglected and a constant PA was maintained until the broken pipe motion resulted in a 2.0A break.

In summary, the geometric piping was modeled using the ANSYS code, and restraints restricting motion were neglected. The state of stress prior to the pipe break was represented at operating temperatures and pressure and included the effects of deadweight. The break was assumed to occur instantaneously, and the break forces were applied conservatively without allowing them to decay as a function of time. Thus, the thermal hydraulics are not required to provide the initial estimate of pipe whip response. For these conditions the displacements of the broken ends of the pipe were tracked and used to compute BOA versus BOT curves which represent the most rapid opening rates physically possible for the piping configuration being addressed.

Figures 2.6-1 through 2.6-4 represent the generic BOA versus BOT curves for the hot and cold leg skirt-supported configurations and the hot and cold leg nozzle-supported configurations.

These generic curves now represent the initial input for the mass and energy calculations and the RV internals asymmetric pressure calculations. These BOA-BOT curves were input into the CRAFT code up to the break opening area they represented so that they reflected a realistic opening rate up to the final opening area.<sup>1</sup> Table 2.6-1 shows the catalog of mass and energy along with the peak generic time for the assumed opening area. Table 2.6-2 provides the catalog of the RV internals asymmetric pressures along with the peak generic opening time for the assumed opening area. Table 2.6-3 lists the actual opening areas and times as well as the as-analyzed areas and times from the catalog in Tables 2.6-1 and 2.6-2. Each of the three distinct as-analyzed phenomena is

tabulated as follows; internal  $\Delta P$ , cavity  $\Delta P$ , and core bounce  $\Delta P$  values. In each case the as-analyzed condition was equal to or worse than the actual condition.

The approach above was necessitated by requiring parallel tasks rather than series tasks. The latter is technically pure, while the former permits overview of the analysis and results in some conservatism in the results.

Table 2.6-1. Spectrum of BOAs/BOTs Used for M&E and Cavity Pressure

<u>Broken leg</u>	<u>Assumed BOA</u>	<u>BOT, s<sup>a</sup></u>
<u>Skirt-Supported Plant</u>		
Hot leg	0.3A	0.012
	0.6A	0.018
	1.0A	0.022
	1.5A	0.028
	2.0A	0.033
Cold leg	0.3A	0.011
	0.6A	0.016
	1.0A	0.020
	1.5A	0.024
	2.0A	0.027
<u>Nozzle-Supported Plant</u>		
Hot leg	1.02A	0.019
Cold leg	0.25A	0.005

<sup>a</sup>BCT: time to open to air ..

Table 2.6-2. Internal Differential Pressure Catalog

<u>Broken leg</u>	<u>Assumed BOA</u>	<u>BOT, s<sup>a</sup></u>
<u>Skirt-Supported Plant</u>		
Hot leg in RV cavity	0.3A	0.012
	0.6A	0.018
	1.0A	0.022
	1.5A	0.028
	2.0A	0.033
Hot leg at elbow	1.0A	0.022
	2.0A	0.033
Cold leg in RV cavity	0.3A	0.011
	0.6A	0.016
	1.0A	0.020
	1.5A	0.024
	2.0A	0.027
Cold leg at elbow	1.0A	0.020
	2.0A	0.027
<u>Broken leg</u>	<u>Actual BOA</u>	<u>BOT, s<sup>a</sup></u>
<u>Nozzle-Supported Plant</u>		
Hot leg in RV cavity	1.024A	0.064
Hot leg at elbow	1.033A	0.057
Cold leg in RV cav	0.242A	0.070
Cold leg at elbow	1.167A	0.043

<sup>a</sup>BOT: time to open to area.

Table 2.6-3. Analyzed Opening, Area/Time to Area

	Actual opening, area/time to peak, A/s	Analyzed opening, area/time to area		
		Internals, A/s	Cavity $\Delta P$ , A/s	Core bounce $\Delta P$ , A/s
<u>Oconee</u>				
Hot leg RV nozzle elbow	0.52/0.027	0.60/0.018	0.60/0.018	2.00/0.033
	0.45/0.020	1.00/0.022	--	2.00/0.033
Cold leg RV nozzle elbow	2.00/0.027	2.00/0.027	2.00/0.027	2.00/0.027
	2.00/0.027	2.00/0.027	--	2.00/0.027
<u>TMI-1</u>				
Hot leg RV nozzle elbow	1.39/0.045	1.50/0.028	1.50/0.028	2.00/0.033
	1.04/0.033	2.00/0.033	--	2.00/0.033
Cold leg RV nozzle elbow	2.00/0.027	2.00/0.027	2.00/0.027	2.00/0.027
	2.00/0.027	2.00/0.027	--	2.00/0.027
<u>CR-3</u>				
Hot leg RV nozzle elbow	1.17/0.046	1.50/0.028	1.50/0.028	2.00/0.033
	0.85/0.033	2.00/0.033	--	2.00/0.033
Cold leg RV nozzle elbow	2.00/0.027	2.00/0.027	2.00/0.027	2.00/0.027
	2.00/0.027	2.00/0.027	--	2.00/0.027
<u>ANG-1</u>				
Hot leg RV nozzle elbow	0.38/0.030	0.60/0.018	0.60/0.018	2.00/0.033
	0.27/0.018	1.00/0.022	--	2.00/0.033
Cold leg RV nozzle elbow	2.00/0.027	2.00/0.027	2.00/0.027	2.00/0.027
	2.00/0.027	2.00/0.027	--	2.00/0.027

2.6-6

Figure 2.6-1. Hot Leg Guillotine Break at Reactor Vessel Outlet - Unrestrained - Generic Skirt-Supported Reactor

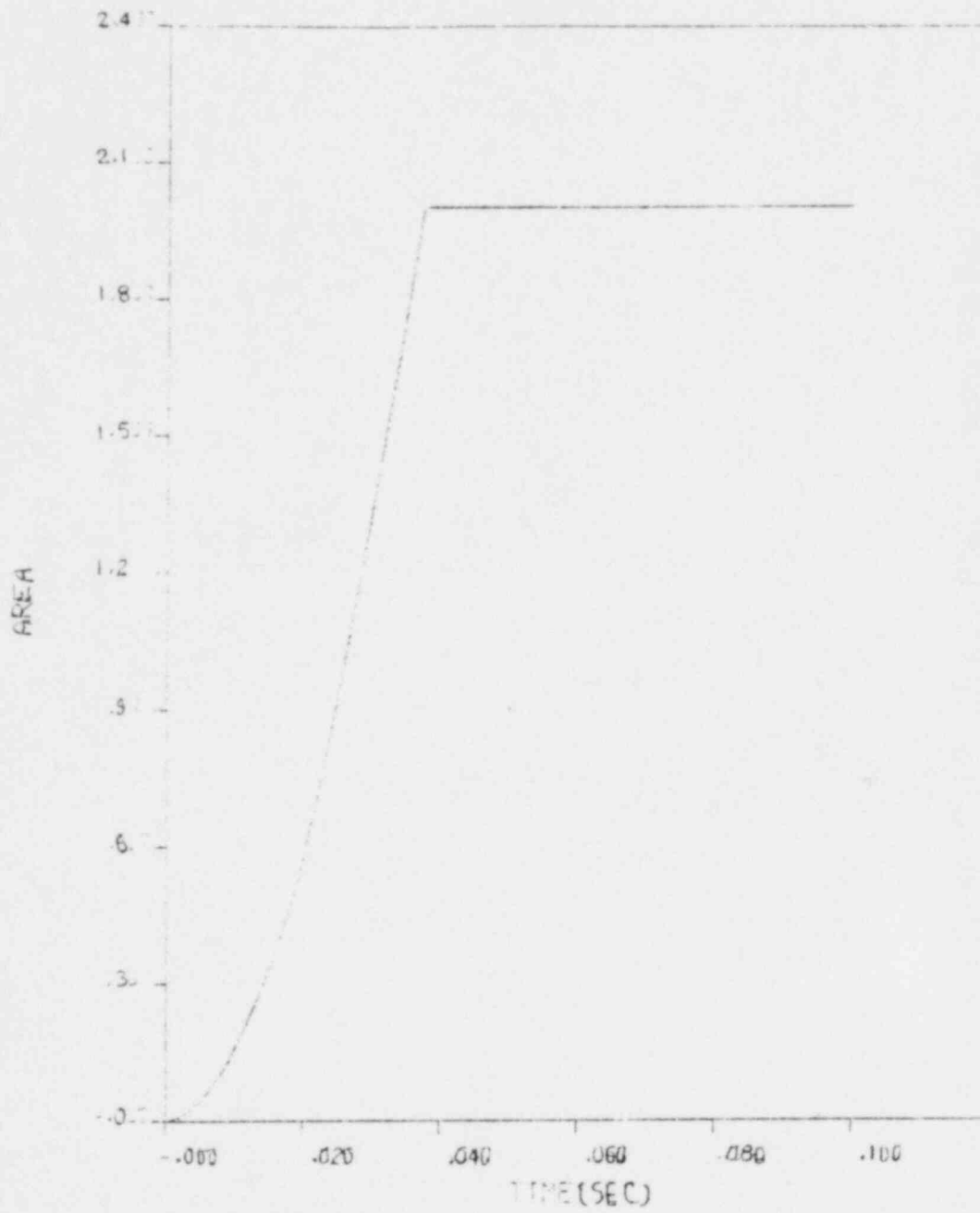


Table 2.6-3. (Cont'd)

	Actual opening, area/time to peak, A/s	Analyzed opening, area/time to area		
		Internals, A/s	Cavity $\Delta P$ , A/s	Core bounce $\Delta P$ , A/s
<u>Rancho Seco</u>				
Hot leg RV nozzle elbow	0.74/0.041	1.00/0.022	0.75/0.019	2.00/0.033
	0.55/0.026	1.00/0.022	--	2.00/0.033
Cold leg RV nozzle elbow	2.00/0.027	2.00/0.027	2.00/0.027	2.00/0.027
	2.00/0.027	2.00/0.027	--	2.00/0.027
<u>DB-1</u>				
Hot leg RV nozzle elbow	1.02/0.064	1.02/0.064	1.02/0.019	1.02/0.064
	1.03/0.057	1.03/0.057	--	1.03/0.057
Cold leg RV nozzle elbow	0.25/0.100	0.25/0.070	0.25/0.005	0.25/0.070
	1.17/0.049	1.17/0.048	--	1.17/0.048
<u>TMI-2</u>				
Hot leg RV nozzle elbow	--	2.00/0.033	2.00/0.033	2.00/0.033
	--	2.00/0.033	--	2.00/0.033
Cold leg RV nozzle elbow	--	2.00/0.027	2.00/0.027	2.00/0.027
	--	2.00/0.027	--	2.00/0.027

2.6-7

Figure 2.6-2. Cold Leg Guillotine Break at Reactor Vessel – Unrestrained – Generic Skirt-Supported Reactor

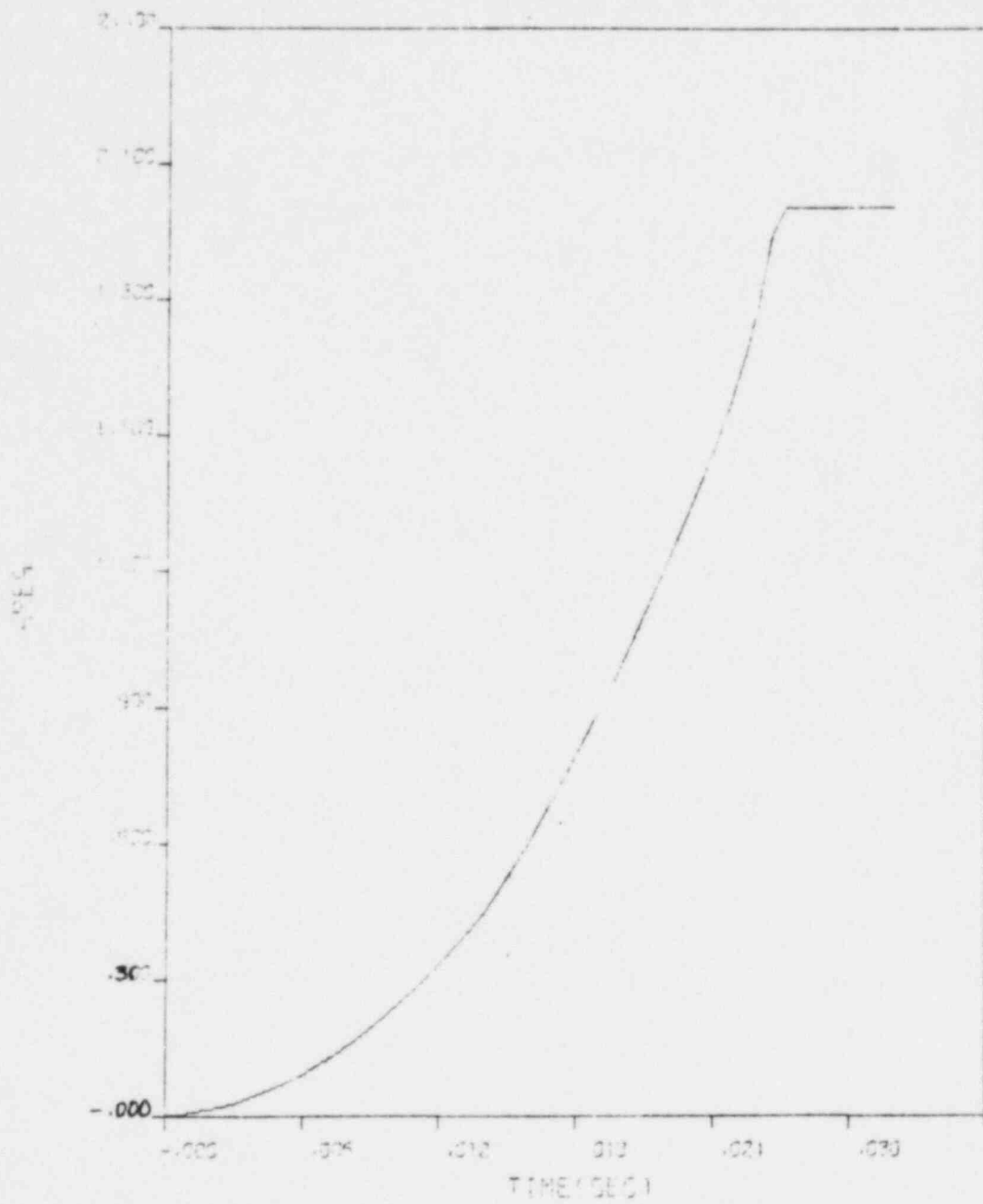




Figure 2.6-3. Hot Leg Guillotine Break — Unrestrained —  
Generic Nozzle-Supported Reactor

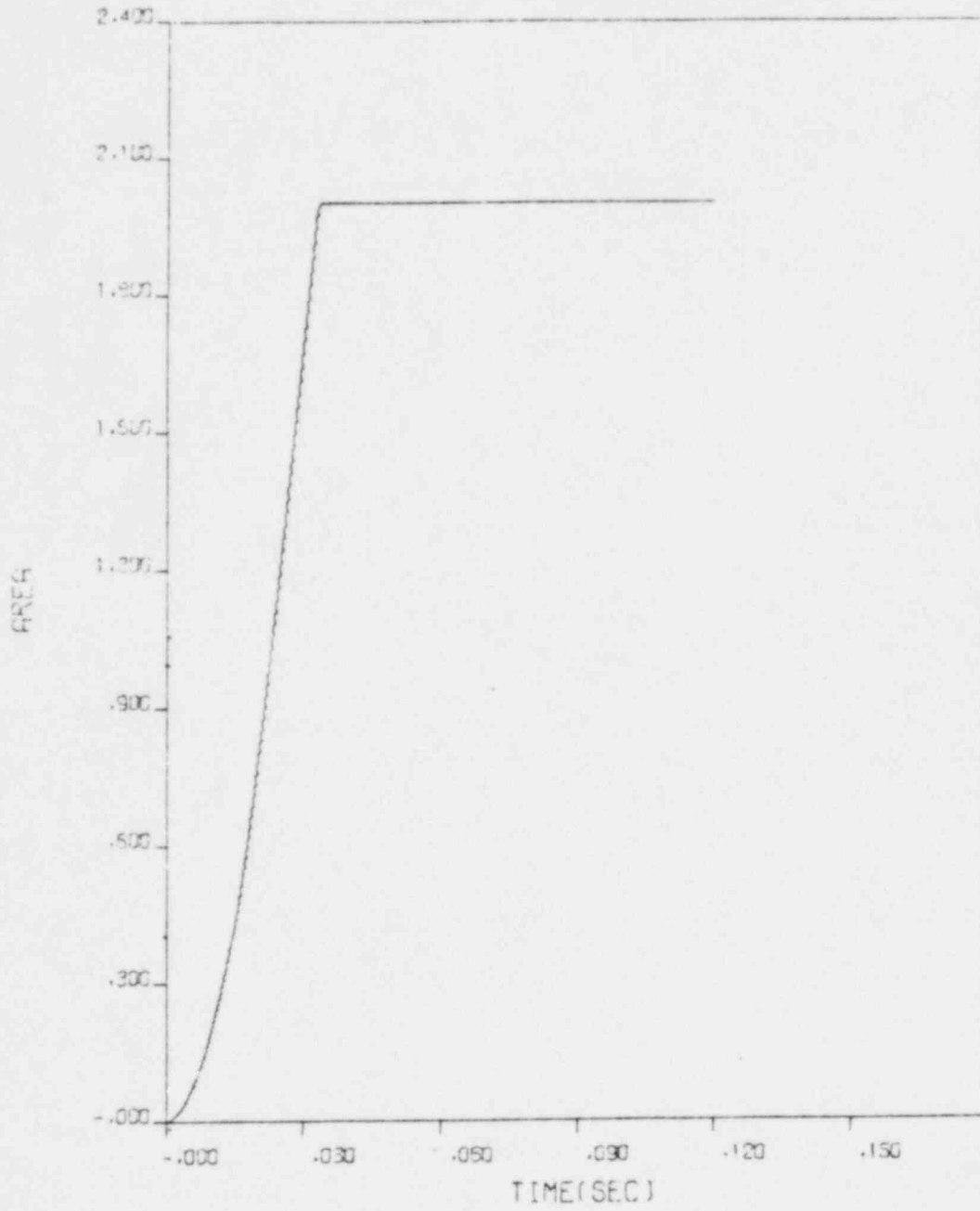
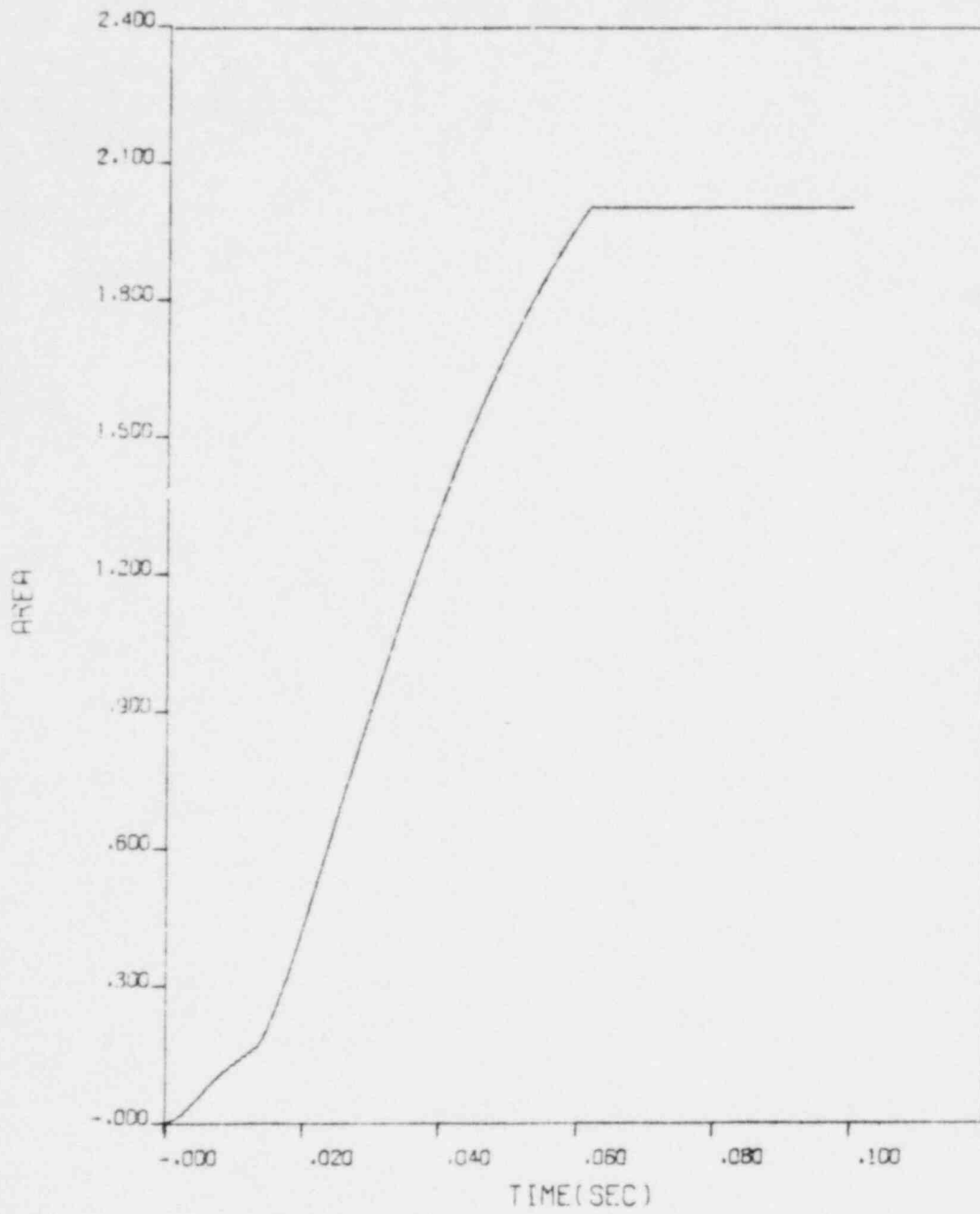


Figure 2.6-4. Upper Cold Leg Guillotine Break — Unrestrained —  
Generic Nozzle-Supported Reactor



2.7. In section 4.4, are the break locations analyzed depicted in Figure 4.1-1? Clarify the break locations specified in section 4.4 (first paragraph, last sentence).

Response

Break locations analyzed are depicted in Figure 4.1-1.<sup>2</sup> Pipe break locations were selected as specified in section 4.1.<sup>2</sup> Hot and cold leg guillotine pipe breaks were considered at the terminal ends of the reactor vessel nozzles, at the entrance to the lower hot elbow, and at the cold leg elbow exit as illustrated in Figure 4.1-1.

2.8. In section 8.2, what are the assumed break locations, areas, and times?

Response

Break locations are shown in Figure 4.1-1<sup>2</sup> and are either at the reactor vessel nozzle or at the first elbow from the RV, which is inside the steam generator cavity. For the skirt-supported plants a spectrum of break sizes (2.0A, 1.5A, 1.0A, 0.6A, and 0.3A) for guillotine breaks at the reactor vessel inlet and outlet nozzles was analyzed. The skirt-supported plants were also analyzed for a 2.0A and a 1.0A break inside the steam generator cavity. Table 8.2-3<sup>2</sup> presents data for the nozzle-supported plant, Davis-Besse 1, and is for the actual BOA and generic BOT given in Table 2.6-3 of the response to question 2.6. BOTs for the skirt-supported plants are based on a generic break opening schedule, and no pipe restraint effects are included for this analysis. Therefore, the assumed BOT for each BOA is provided in Table 2.6-3.

2.9. In section 4.4, clarify the last sentence in the second paragraph. Identify or provide those BOA curves for each break location considered. Are Figures 4.2-3 through 4.2-17 generic or actual BOA curves? The stated sentence appears to conflict with use of actual BOA codes mentioned in the first paragraph for the nozzle-supported plant (please clarify)

Response

For each of the postulated breaks, the generic BOA versus time data were used and are provided in the response to question 2.6. Davis Besse 1, the nozzle-supported plant, used actual BOAs for each of the postulated breaks but used a generic break opening time based on an unrestrained guillotine break and the time to reach the restrained BOA (generic curve provided in response to question 2.6). Figures 4.2-3 through 4.2-17 of reference 2 are actual BOA curves.

2.10. In Appendix C, paragraph 2, how were the "generic fluid conditions" established?

Response

A generic set of fluid conditions was established so that the most conservative internal loads would be produced. Knowing that the magnitudes of reactor internal loads are directly proportional to the coldest fluid temperatures (lowest T) between inlet and outlet nozzles, the various parameters for the skirt-supported plants given in Table 2.10-1 were compared. The various parameters were taken from heat balance data sheets, and the design total NSS power levels were verified as those being in the SAR. Three cases (Consumers not included) were evaluated to determine which generic fluid conditions tabulated in Appendix C of reference 2 for the skirt-supported plants gave the most conservative (highest) internal loadings.

Table 2.10-1. Comparison of Plant Parameters

Design criteria	Case 1 <sup>a</sup>	CR-3	Case 2 <sup>a</sup>	Case 3 <sup>a</sup>
Core power, MWt	2552	2544	2552	2772
NSS power, MWt	2568	2560	2568	2789
RV inlet temp, F	554	554.4	554	555.7
RV outlet temp, F	604	603.6	604	608.6
Reactor avg temp, F	579	579.0	579	582
Reactor flow, 10 <sup>6</sup> lbm/h	131.32	131.4	131.5	131.3
SG outlet temp, F	553.7	554.0	553.7	555.4
Steam outlet temp, F	570	570.2	558	585
Steam outlet pressure	925	925	928	925
FW temp, F	460	459	433	455
FW pressure, psi	1000	1000	1003	1000
FW flow, 10 <sup>6</sup> lbm/h	11.2	11.0	10.75	11.76
Steam superheat	35	35.0	22.4	50
RC pump power, MWt	16	16	16	17
RC pump/flow, gpm	88,000	88,000	88,000	88,000

<sup>a</sup>Case 1: Oconee 1, 2, and 3; ANO-1; and TMI-2.  
Case 2: Consumers.  
Case 3: Davis-Besse 1, TMI-1, and Rancho Seco.

2.11. In Appendix C, for the skirt-supported plant, there are more downcomer nodes than in the "audit model" in BAW-10132P-A. Discuss this and any other differences and how it will impact RV internal pressures and loads.

Response

There are two main differences between the skirt-supported model (Appendix C of reference 2) and the audit model (BAW-10132P-A, Figures 3-25 through 3-30<sup>6</sup>):

- A 12-circumferential node downcomer in the skirt-supported model compared to an 8-node downcomer in the audit model.
- A 10-node upper level plenum annulus in the skirt-supported model compared to a 4-node representation in the audit model.

Noting that the audit model in BAW-10132P-A was shown to provide sufficient noding detail, the increase in noding detail in the skirt-supported model provides for a slightly improved representation of the eight vent valves. In addition, as with all CRAFT2 type codes (RELAP, for example), as modeling detail increases, the average node/flow path properties more closely approximate local conditions.

The impact of added detail in the skirt-supported model is deemed a minor but favorable model improvement in that it more closely approximates the actual fluid system.



2.12. In BAW-10132P-A, Supplement 1, Figure D-1 (design model, 16-node downcomer) shows a peak differential pressure 290 psi less than Figure A-1 (audit model, 40-node downcomer). Based on this indication of lower differential pressures when modeling about 50% fewer downcomer nodes, justify use of the 20-node CRAFT2 downcomer model for the nozzle-supported plant to evaluate RV internals pressures and loads.

Response

Comparing Figures A-1 and D-1 in BAW-10132P-A, Supplement 1<sup>7</sup>, is not valid for this application. The legends associated with each of these figures clearly show that different vent valve simulations were employed in the hydraulic modeling. Figure A-1 (40-node downcomer) stems from a model using dynamic vent valves, while Figure D-1 (16-node downcomer) is based on a model with instantaneously opening vent valves.

A proper downcomer noding comparison is presented in Table 3-10 of BAW-10132P-A.<sup>6</sup> This table clearly demonstrates that using a coarse downcomer nodal representation is conservative and that dynamic vent valves are realistic.

- 2.13. For the skirt-supported plant for a cold leg RV nozzle break, show a sample calculation for integrating the pressures around the core barrel periphery to get a directional force per unit length. Consider volume nodes 102 through 113 (at one elevation only) in Figures C-2 and C-3.

Response

As agreed at the April 2, 1981, meeting, methods and equations are provided for integrating the pressure around the core barrel periphery.

The equations used for force calculation of resultant horizontal components  $F_{xj}$  and  $F_{zj}$  (see Figure 2.13-1) are as follows:

$$F_{xj} = \sum_i A_i P_i \cos \alpha_i = \sum_i F_i \cos \alpha_i$$

$$F_{zj} = \sum_i A_i P_i \sin \alpha_i = \sum_i F_i \sin \alpha$$

where

$A_i$  = projected area of node  $i = l_i h_i$ ,

$P_i$  = absolute pressure of node  $i$ , assume uniform in node  $i$ ,

$F_i$  = resultant force of node  $i = A_i P_i$ ,

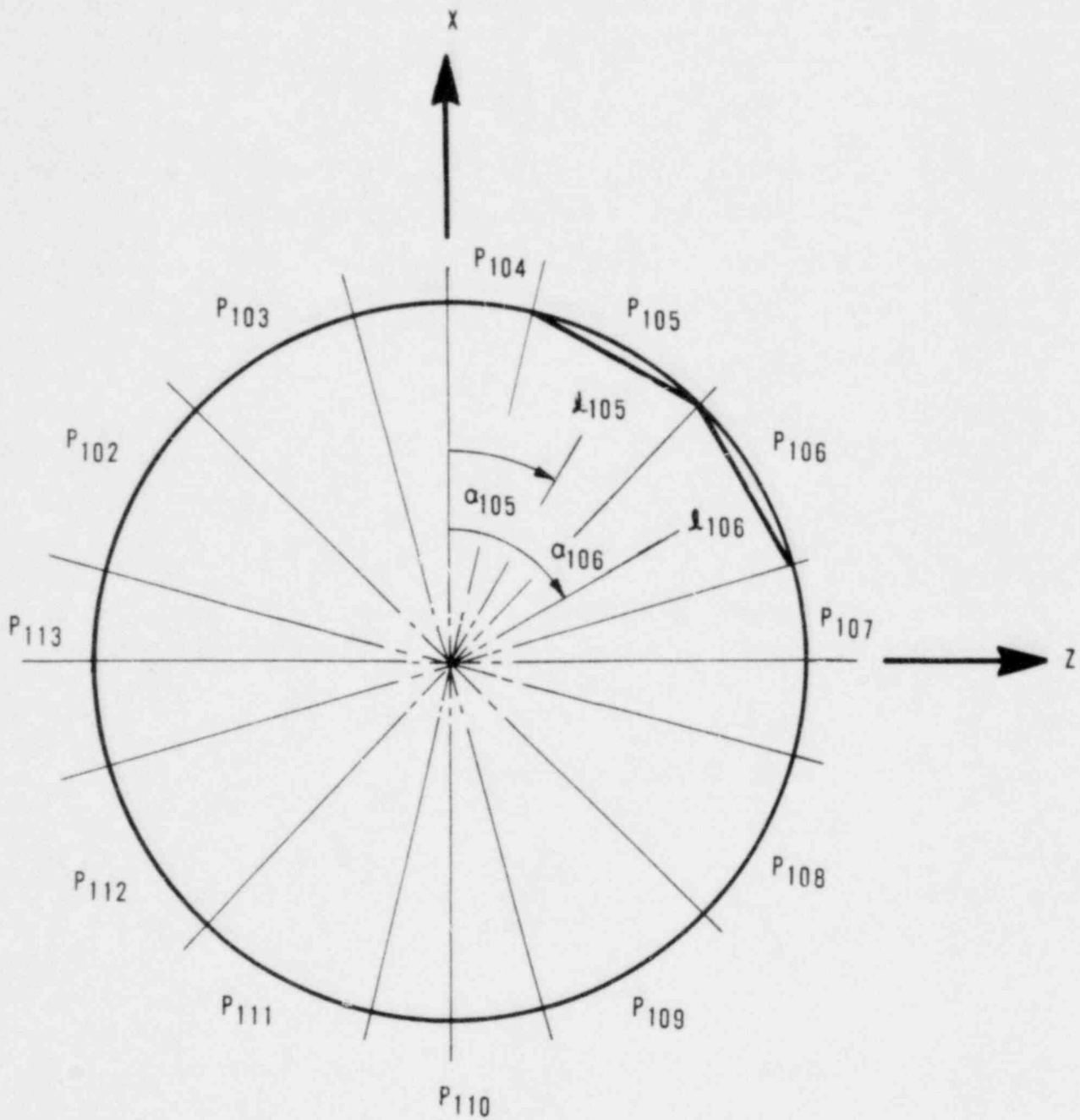
$\alpha_i$  = orientation angle of  $F_i$  with respect to X axis,

$l_i$  = projected length of node  $i$ ,

$h_i$  = projected height of node  $i$ .

The resultant horizontal force is applied at the center of the elevation of the projected area.

Figure 2.13-1. Circumferential Pressure/Node Relationship for Skirt-Supported Plant (Figure C-2<sup>2</sup>)



- 2.14. What effect could a sticking vent valve have, i.e., late opening/closing (or) failure to open/close, relative to RV internals differential pressures for hot and cold leg breaks?

Response

Internals vent valves are installed in the core support shield to prevent a pressure buildup that might interfere with core cooling following a postulated inlet pipe rupture. Under all normal operating conditions, the vent valve will be closed. In the event of a pipe rupture in the cold leg of the reactor loop, the valve will open to permit steam generated in the core to flow directly to the leak and will permit the core to be rapidly re-covered and adequately cooled after emergency core coolant has been supplied to the reactor vessel.

The internals vent valves were designed to relieve the pressure generated by steaming in the core following a LOCA so that the core will remain sufficiently cooled. The valves were designed to withstand the forces resulting from rupture of either a reactor coolant inlet or outlet pipe. To verify the structural adequacy of the valves to withstand the pressure forces and perform the venting function, the following tests were performed:

1. A full-size prototype valve assembly (valve disc-retaining mechanism and valve body) was hydrostatically tested to the maximum pressure expected to result during the blowdown.
2. Sufficient tests were conducted at zero pressure to determine the frictional loads in the hinge assembly, the inertia of the valve disc, and the disc rebound resulting from impact of the disc on the seat so that the valve response to cyclic blowdown forces may be determined analytically.
3. A prototype valve assembly was pressurized to determine the pressure differential required to cause the valve disc to begin to open. A determination of the pressure differential required to open the valve disc to its maximum open position was simulated by mechanical means.
4. A prototype valve assembly was successfully installed and removed remotely in a test stand to confirm the adequacy of the vent valve handling tool.
5. A 1/6 scale model valve disc closing force (excluding gravity) test was conducted.

6. The full-size prototype valve's response to vibration was determined experimentally to verify prior analytical results, which indicated that the valve disc would not move relative to the body seal face as a result of vibration caused by transmission of core support shield vibrations. The prototype valve was mounted in a test fixture which duplicated the method of valve mounting in the core support shield. The test fixture with valve installed was attached to a vibration test machine and excited sinusoidally through a range of frequencies which encompassed those that may reasonably be anticipated for the core support shield during reactor operation. The relative motion between the valve disc and seat was monitored and recorded during the test. The results indicated that there was no relative motion of the valve to its seat for conditions simulating operating conditions. After no relative motion was observed or recorded during the test, the valve disc was manually forced open during a test to observe its response. The disc closed with impact on its seat, rebounded open, and re-seated with no adverse effects to valve seal surfaces, characteristics, or performance. From this oscillograph record, the natural frequency of the valve disc was conservatively calculated as approximately 1500 Hz, whereas, the range of frequencies for the primary system (including internals components) has been established as 15 to 160 Hz. These frequencies are separated by an ample margin to conclude that no relative motion between the valve disc and its seat will occur during normal reactor operation.

Each production valve is subjected to tests 2 and 3 above except that no additional analysis is performed in conjunction with test 2.

The valve disc, hinge shaft, shaft journals (bushings), disc journal receptacles, and valve body journal receptacles have been designed to withstand, without failure, the internal and external differential pressure loadings resulting from a loss-of-coolant accident. The valve materials are nondestructively tested and accepted in accordance with the ASME Code, Section III, requirements for Class A vessels as a reference quality level.

During scheduled refueling outages after the reactor vessel head and the internals plenum assembly have been removed, the vent valves are accessible for visual and mechanical inspection. A hook tool is provided to engage with the valve disc exercise lug. With the aid of this tool, the valve discs are manually exercised during each refueling outage to evaluate disc freedom. The hinge design incorporates special features to minimize the possibility of impairment of valve disc motion during its service life. With the aid of the hook tool, the valve disc can be raised for remote visual inspection of the valve body and disc sealing faces to evaluate observed surface irregularities.

Vent valve design and evaluations are documented in topical report BAW-10005, Rev. 1.<sup>6</sup> Reference 9 is an NRC review of B&W operating experience with reactor internals vent valves and states that B&W design and safety analyses need not include a vent valve flow penalty.

Based on these design considerations, subsequent testing, years of excellent valve performance, and the valves being manually exercised during each refueling, the probability of a late vent valve opening/closing (or) failure to open/close in combination with a guillotine pipe break is essentially zero. Therefore, a sticking vent valve has not been considered in this analysis.

2.15. In establishing the generic skirt-supported plant model, what sensitivity studies were conducted among the various skirt-supported plants to define worst-case internals differential pressures relative to any design or as-built differences in internals geometry, vent areas, flow loss coefficients, and CRAFT2 model volume arrangement and flow inertia terms?

Response

Within manufacturing tolerances, the skirt-supported plants are geometrically the same. Hence, no geometric sensitivity studies were required for the generic, skirt-supported, reactor vessel internals  $\Delta P$  model.

2.16. How were local fluid acceleration (crossflow drag) forces accounted for in RPV internals loads evaluation (such as lateral forces on the control rod guide tubes and support columns during a hot leg break)?

Response

CRAFT2 is not the vehicle used by B&W to calculate crossflow drag type loadings on the control rod guide tubes.<sup>2</sup> Such loads are generated using an appropriate drag formula in conjunction with the maximum crossflow velocity. The drag load is then combined with the induced structural guide tube loadings in order to perform a stress analysis of the control rod guide tubes. The drag load and the induced load occur at different times: the combination without time-phasing yielded satisfactory stress levels.



2.17. In Appendix C, qualify and justify the statement in paragraph 2: "A generic set of fluid conditions was established to serve as a conservative representation of the fluid conditions present in the 177-FA lowered-loop plants." Were actual or SAR design-quoted fluid-parameters used?

Response

See the response to question 2-10.

- 2.18. For all skirt-supported plants, provide a table comparing those cognizant geometric, thermal, and hydraulic parameters used to justify generic grouping. What were the main conditions used to establish the similarity between actual plants and the case analyzed for RPV internals and RCS/ECCS hydraulic transients during subcooled blowdown. Include Davis-Besse 1 parameters for comparison with a nozzle supported plant.

Response

Table comparing plant parameters is included in the response to question 2.10. Main conditions considered were geometric, thermal and hydraulic similarity.

2.19. In Figures C-9 and C-11 identify the downcomer volume nodes adjacent to both hot and cold leg breaks and in Figures C-2 and C-7 for a hot leg break.

Response

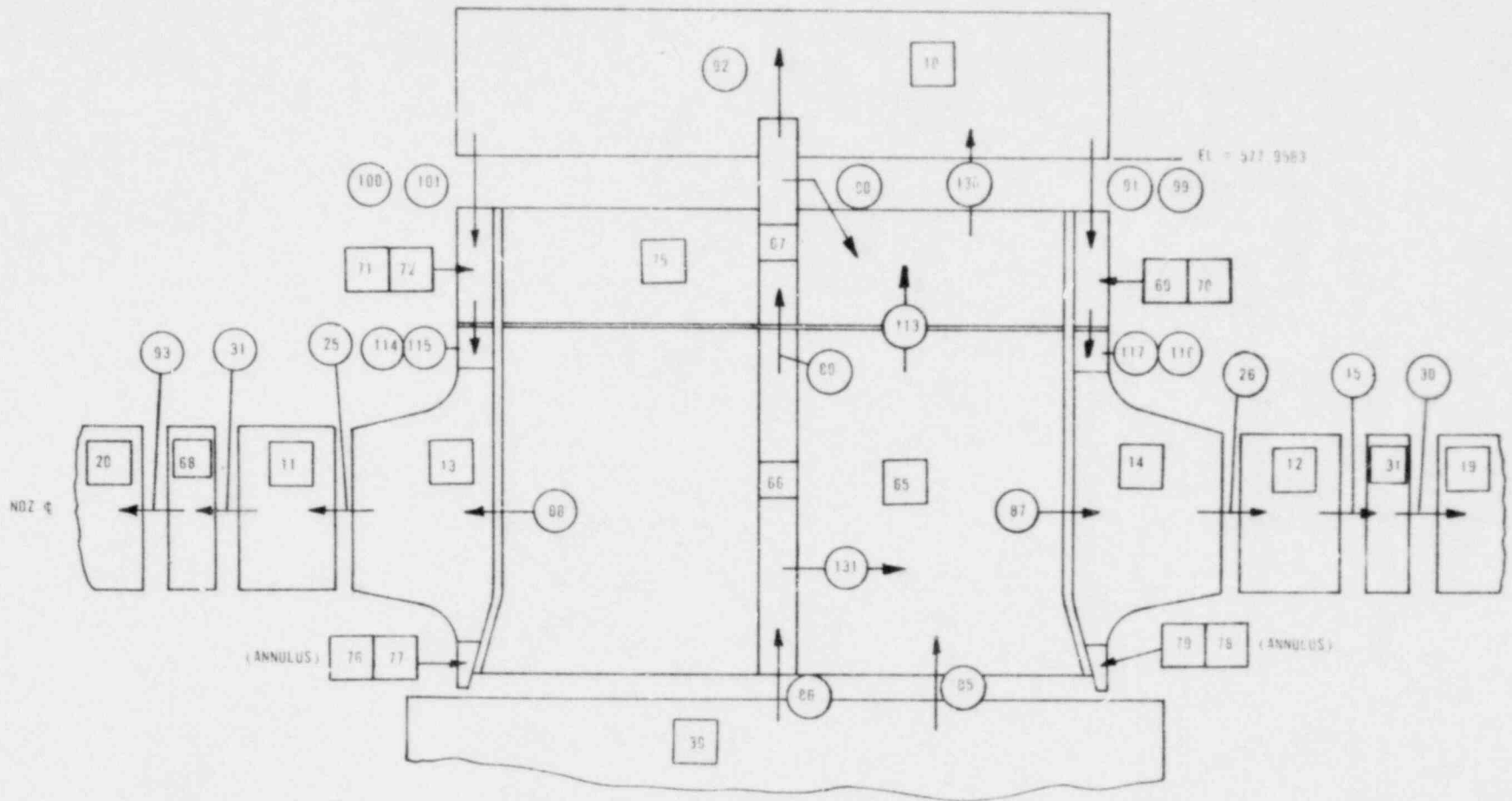
Figures C-1 through C-7<sup>2</sup> show the CRAFT2 noding model for the 17'-FA skirt supported plant. For a hot-leg break, flow would exit the reactor vessel into outlet nozzle control volume 71 shown in Figures C-1 and C-6, via flow path 74 from control volume 95. Figure C-6 shows flow path 250 from control volume 153, flow path 265 from control volume 167, and flow path 264 from control volume 166 flowing into control volume 95. Figure C-2 is an elevation view of the reactor vessel control volumes and does not show broken loop or circumferential volume locations. Figures C-8 through C-11 are the CRAFT2 noding model for the 177'-FA nozzle-supported plant. For a hot leg break, refer to Figures C-11a, -11b, and -11c included with the response to question 2.20. Flow to the break must pass through control volume 14, and the following flow paths feed into volume 14: 87, 103, 104, 116, 117, and 131. For a cold leg break, refer to Figures C-8 through C-10; flow must pass through control volume 26. Control volume 82 connects to control volume 26 via flow path 120. Figure C-9 is an elevation view of the reactor vessel control volumes and does not show broken loop or circumferential volume locations.

2.20. Figure C-11 is not interpretable. Provide a description, with legible volume node numbers, like Figures C-6 and C-7.

Response

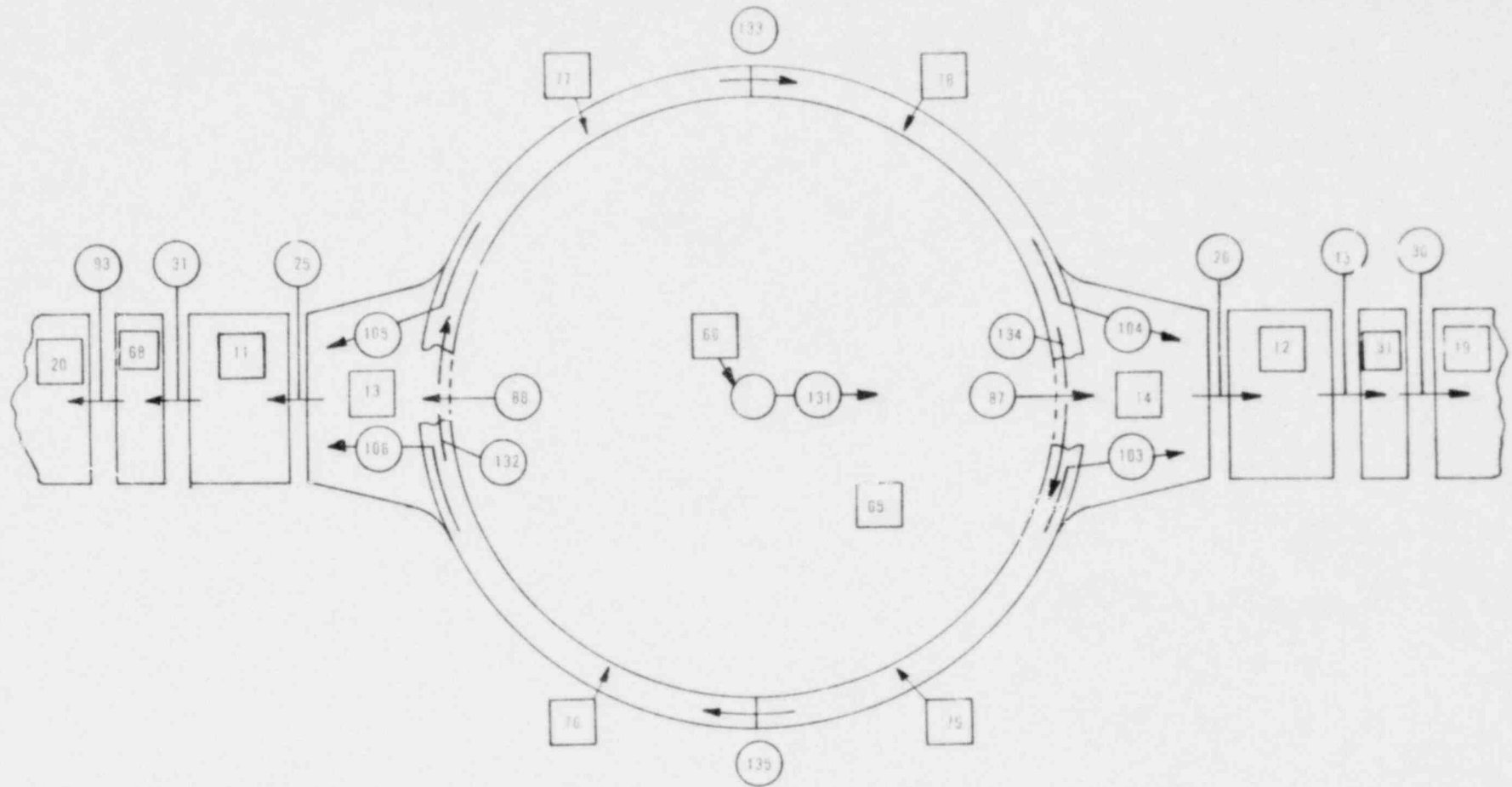
Figure C-11 has been divided into three parts -- Figures 2.20-1, 2.20-2, and 2.20-3 are attached. Description notes have not been provided since the nodes in the outlet annulus are divided into 90° sections.

Figure 2.20-1. CRAFT2 Noding Model for 177-FA Nozzle-Supported Plant - Elevation, Upper Plenum Region



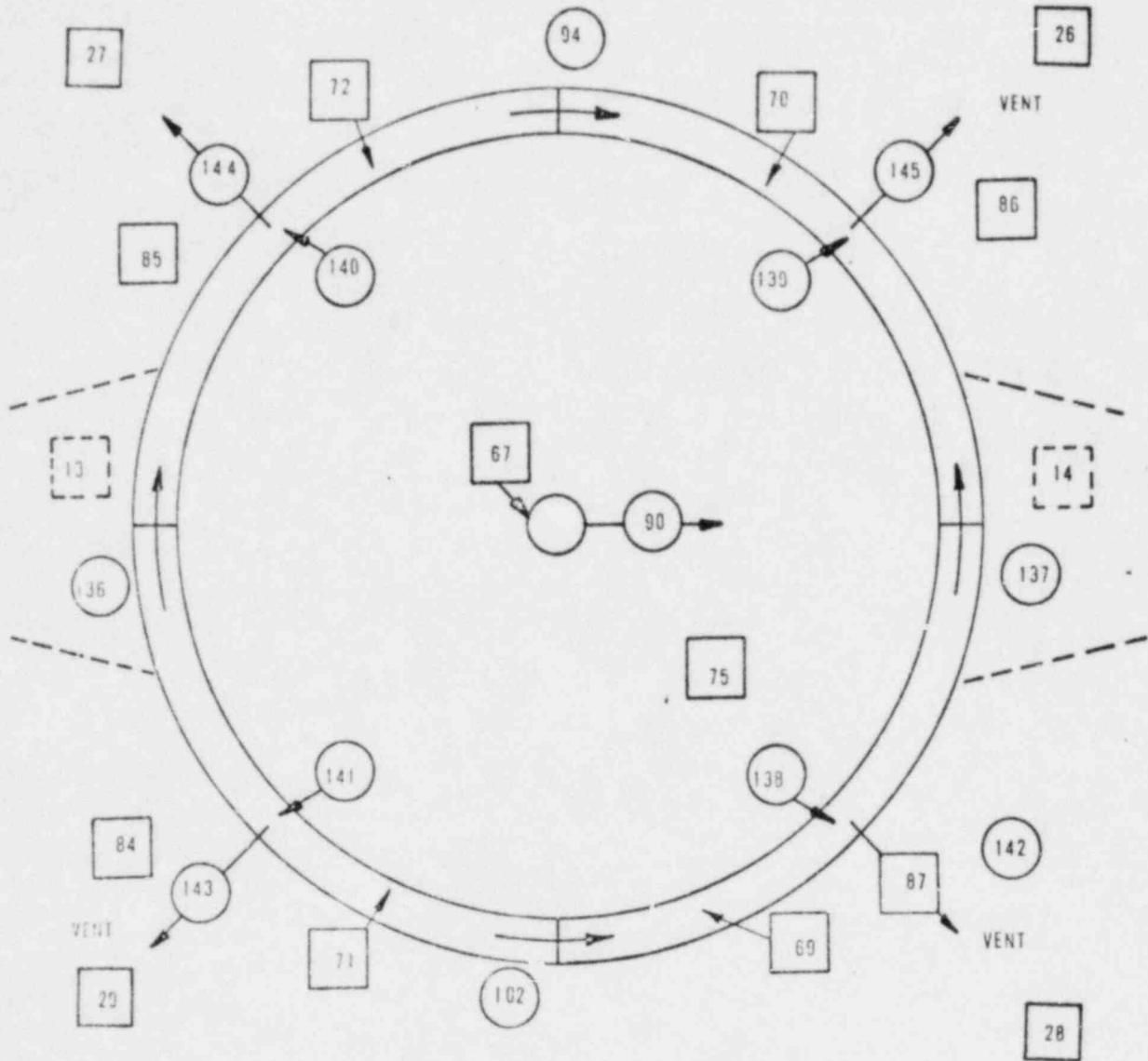
2.20-2

Figure 2.20-2. CRAFT2 Noding Model for 177-FA Nozzle-Supported Plant - Plan, Lower Plenum Region



2.20-3

Figure 2.20-3. CRAFT2 Noding Model for 177-FA Nozzle-Supported Plant - Plan, Upper Plenum Region



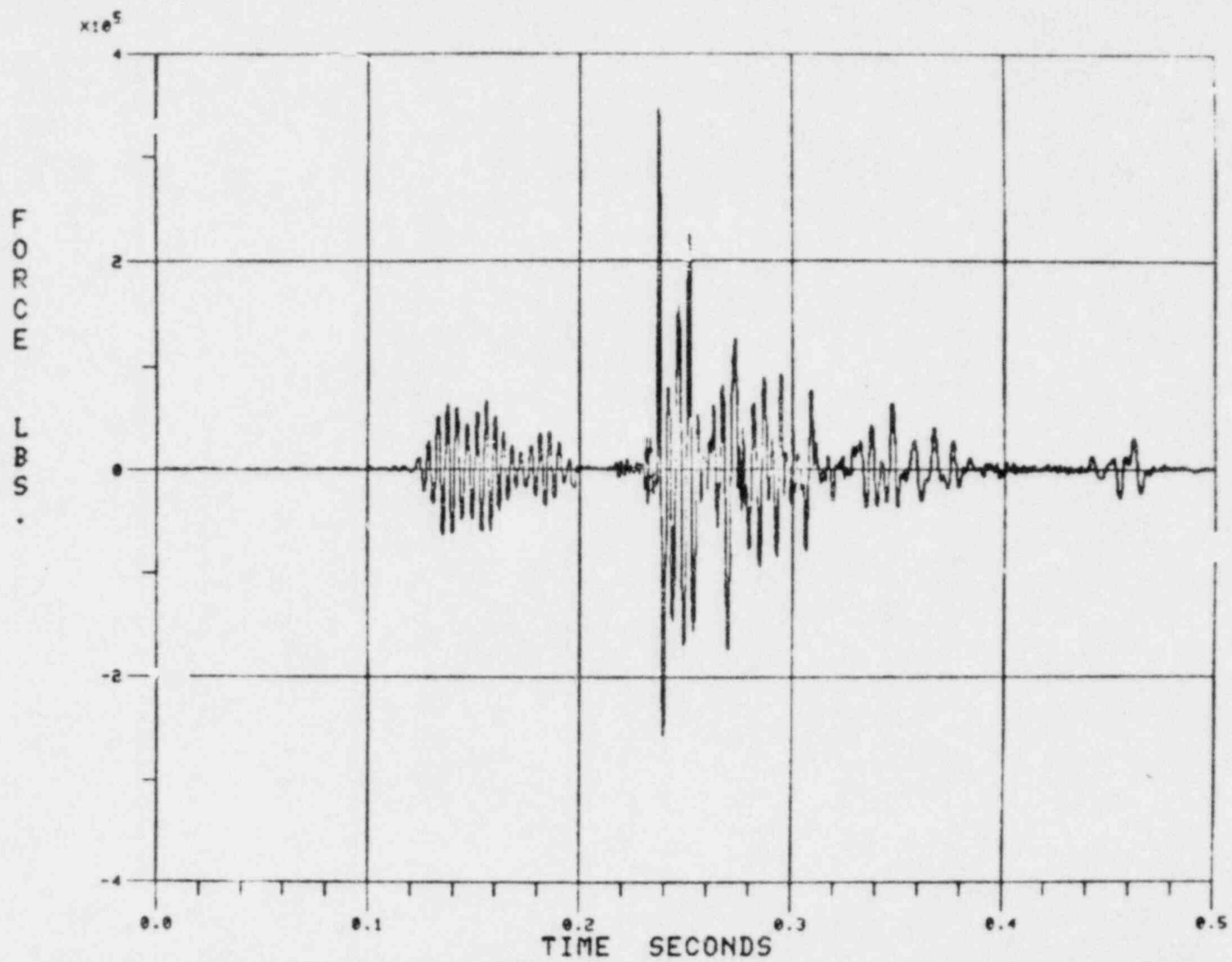
- 2.21. Provide a table of peak differential pressures and loads for each major RV internals component analyzed for both the skirt- and nozzle-supported plants for each BOA and location evaluated. Indicate time of occurrence.

Response

This question was revised at the April 2-3 meeting to provide peak loads across the core support shield and across the fuel assemblies (top to bottom) for reactor vessel inlet and outlet breaks for both skirt- and nozzle-supported plants. Time history forces across the core support shield for 2A hot and cold leg breaks at the RV nozzle are shown in Figures 2.21-1 and 2.21-2, respectively for a skirt-supported plant. Nozzle-supported plant time history forces are shown in Figures 2.21-3 and 2.21-4 for a 1.024A hot leg break and a 0.242A cold leg break at the RV, respectively. Figures 2.21-5 and 2.21-6 show the total force on the core for a 2A hot and cold leg break at the RV, respectively, for a skirt-supported plant. Forces on the core for a nozzle-supported plant are shown in Figures 2.21-7 and 2.21-8 for a 1.024A hot leg and a 0.242A cold leg break at the RV, respectively. Joint numbers in these figures refer to Figure D-5 of reference 2.



Figure 2.21-1. Force Across Core Support Shield - 2A Hot Leg Break at RV, Skirt Supported Plant



2.21-2

Figure 2.21-1. (Cont'd)

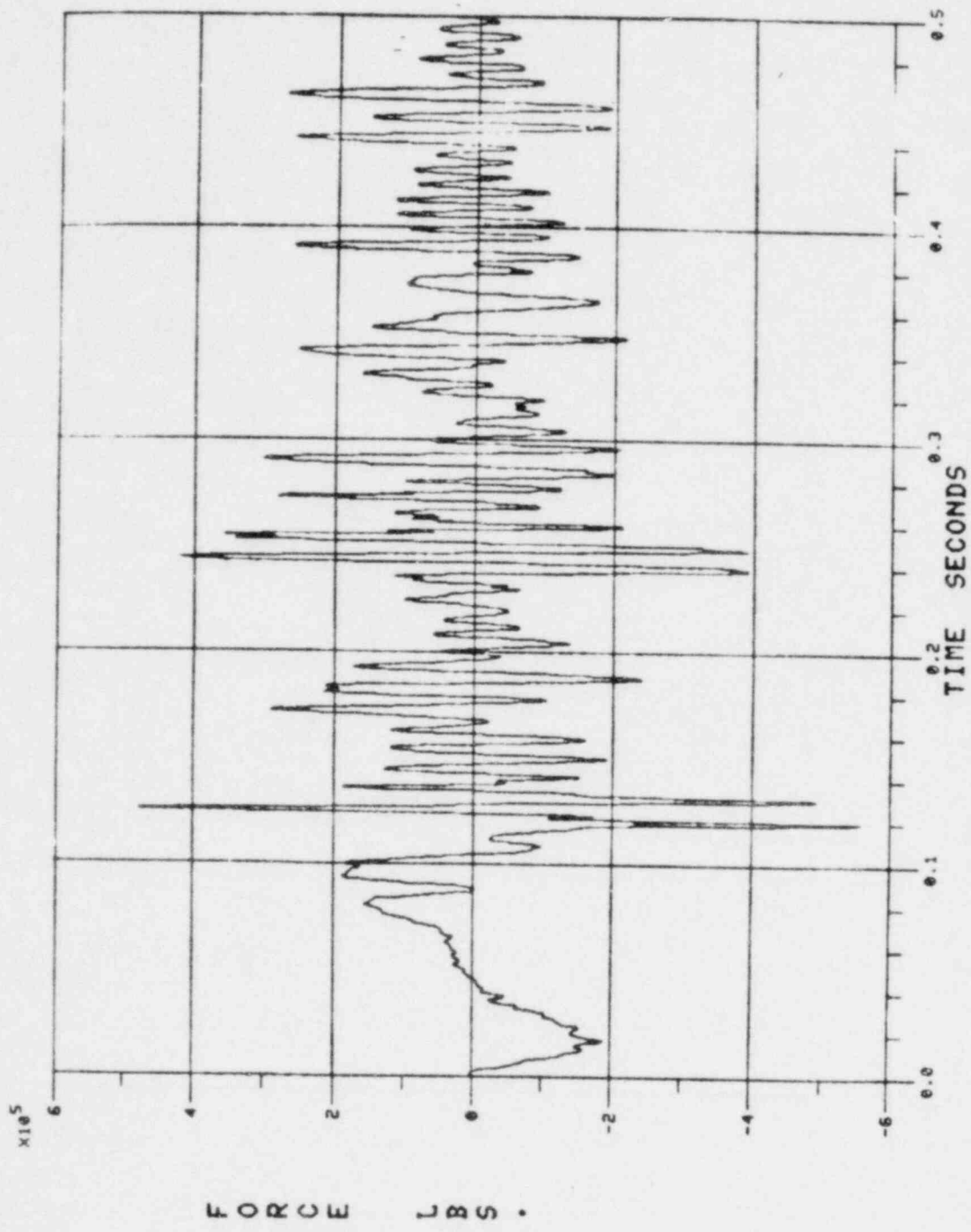
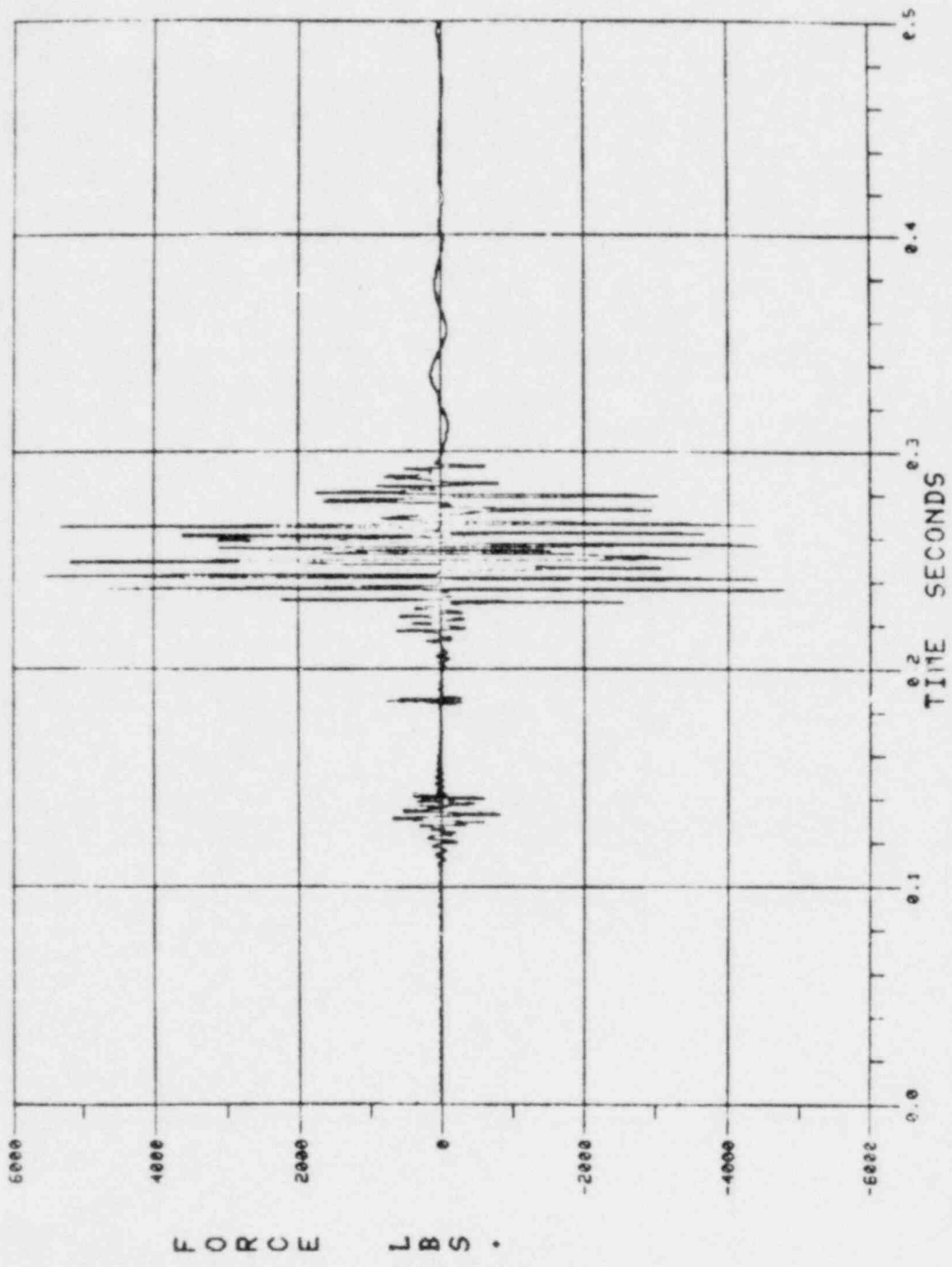
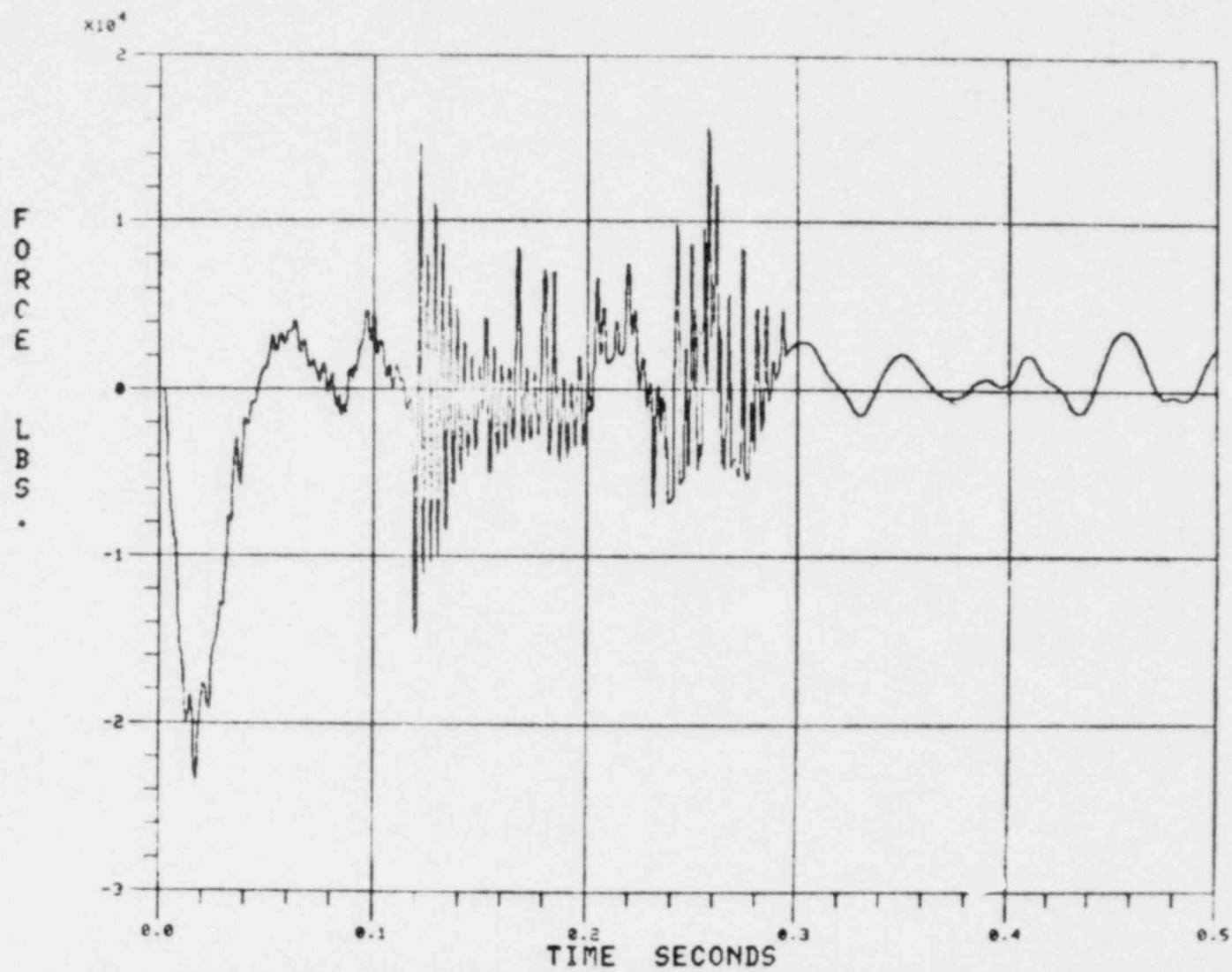


Figure 2.21-1. (Cont'd)



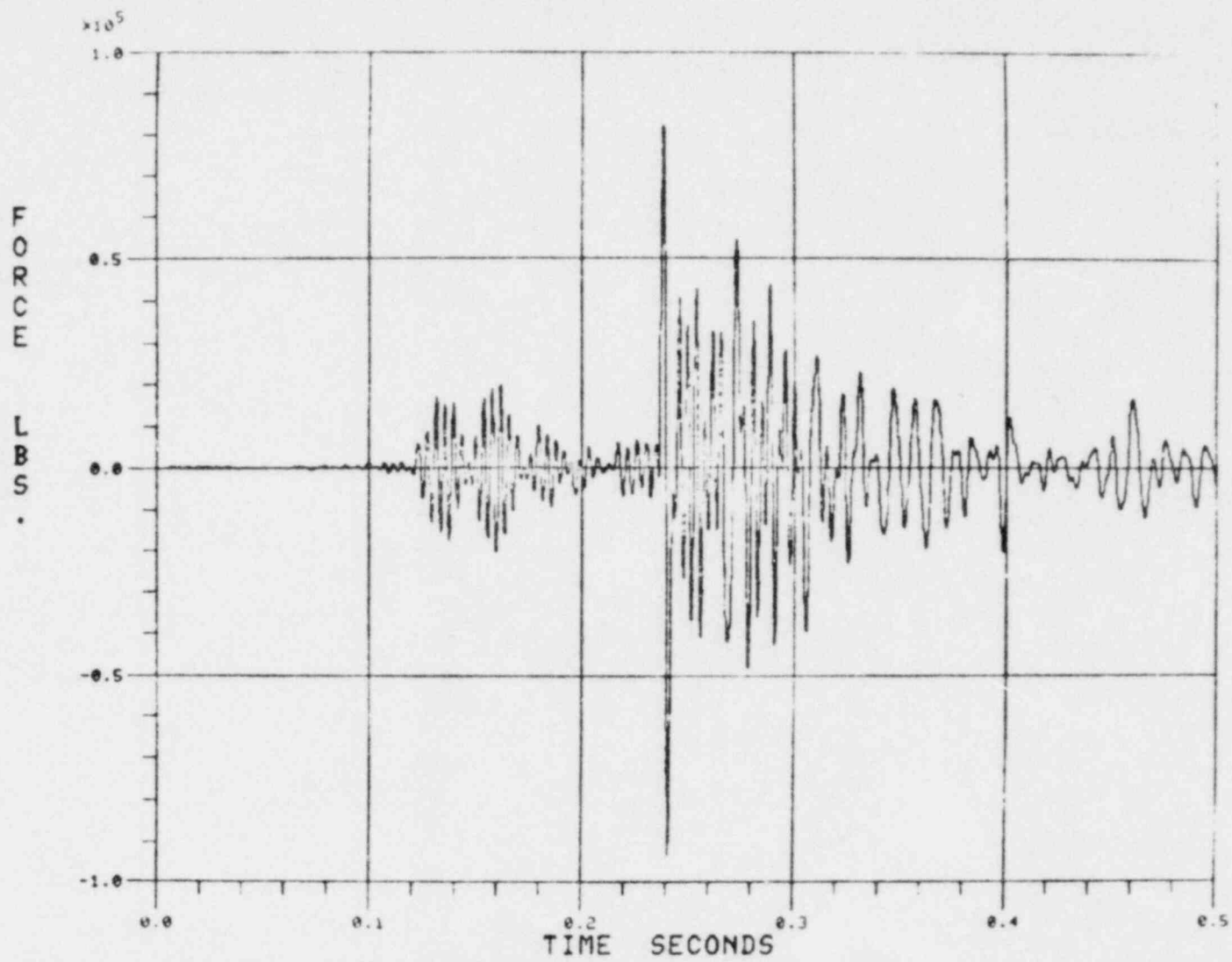
JT 53 FX 37

Figure 2.21-1. (Cont'd)



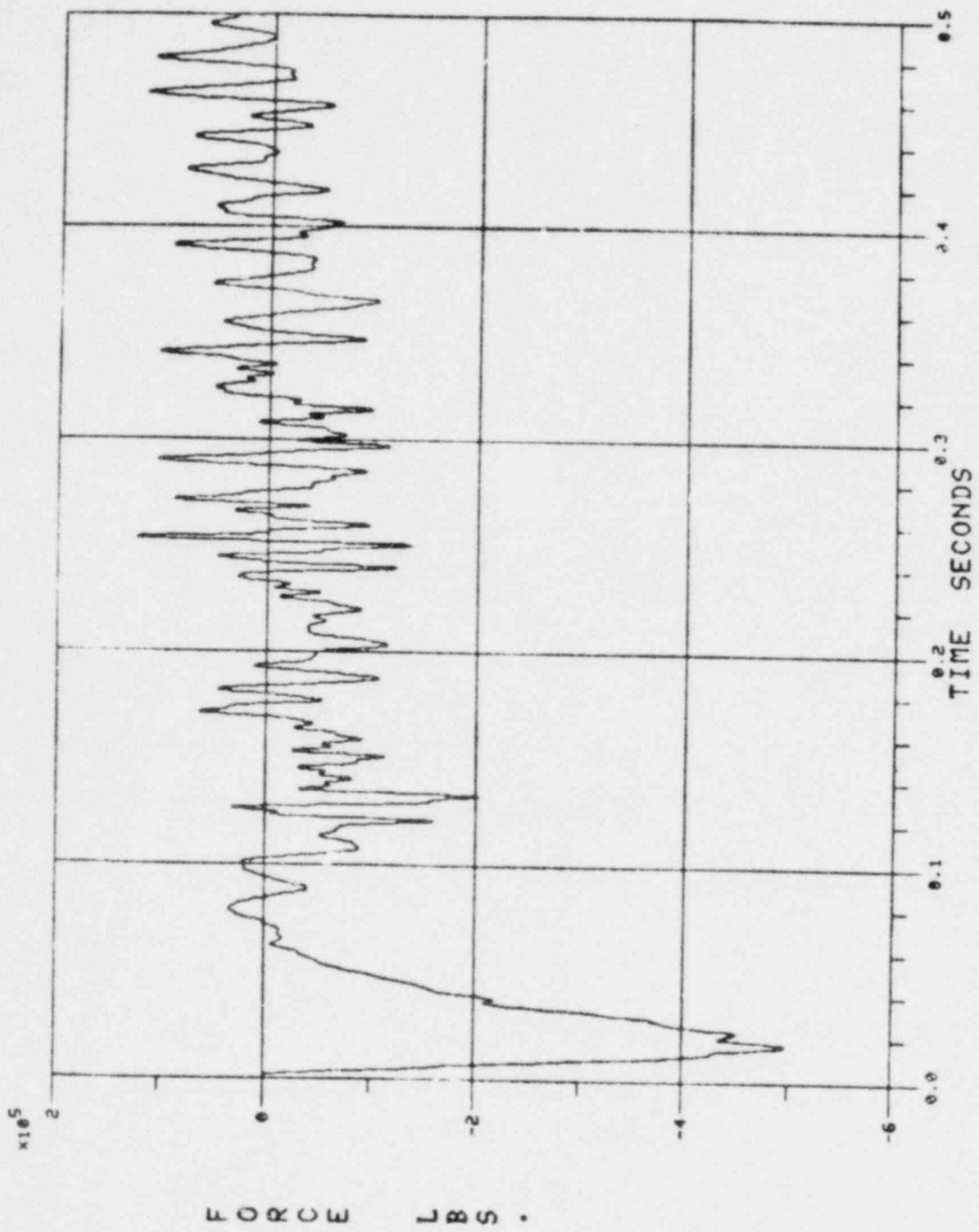
2.21-5

Figure 2.21-1. (Cont'd)



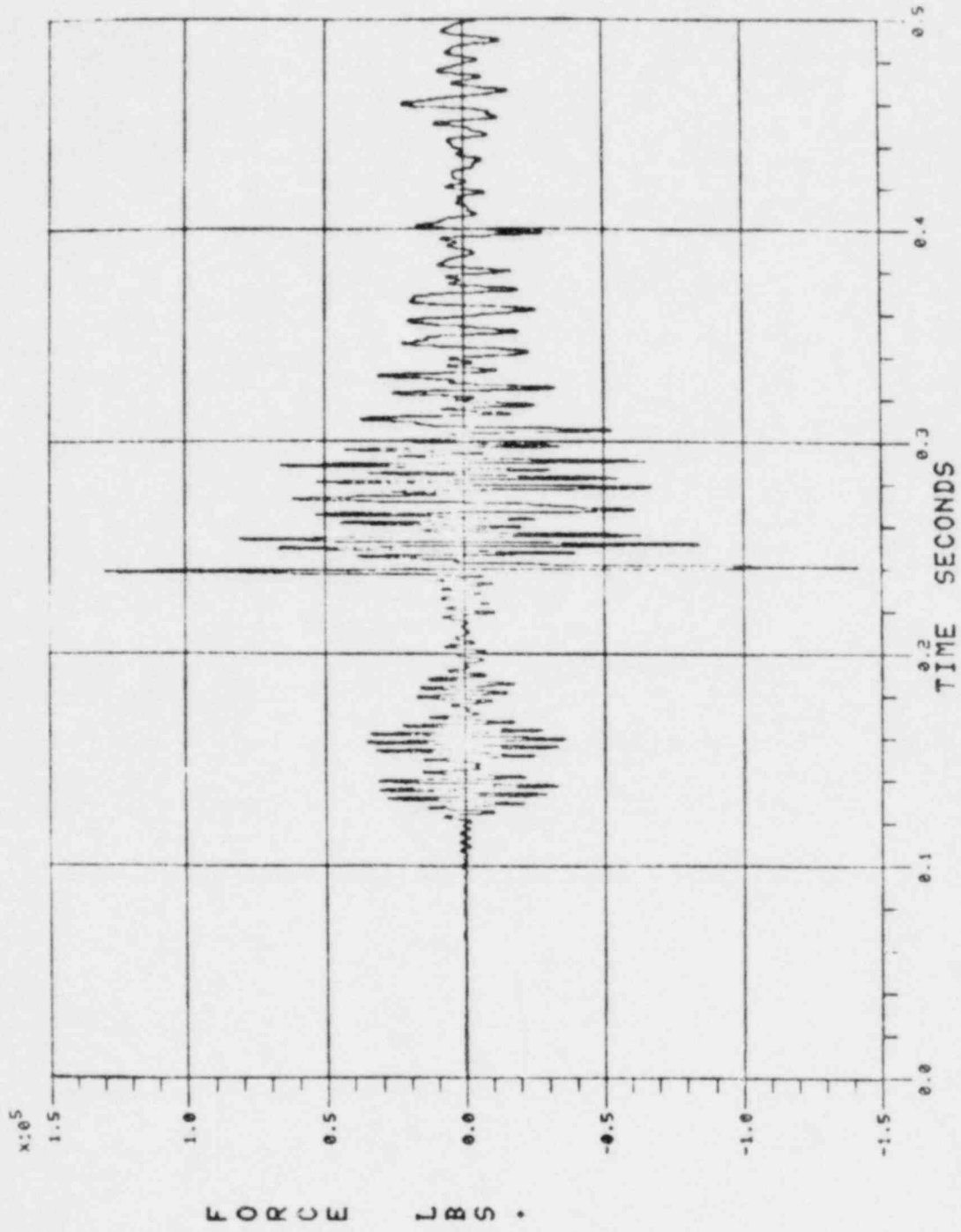
2.21-6

Figure 2.21-1. (Cont'd)



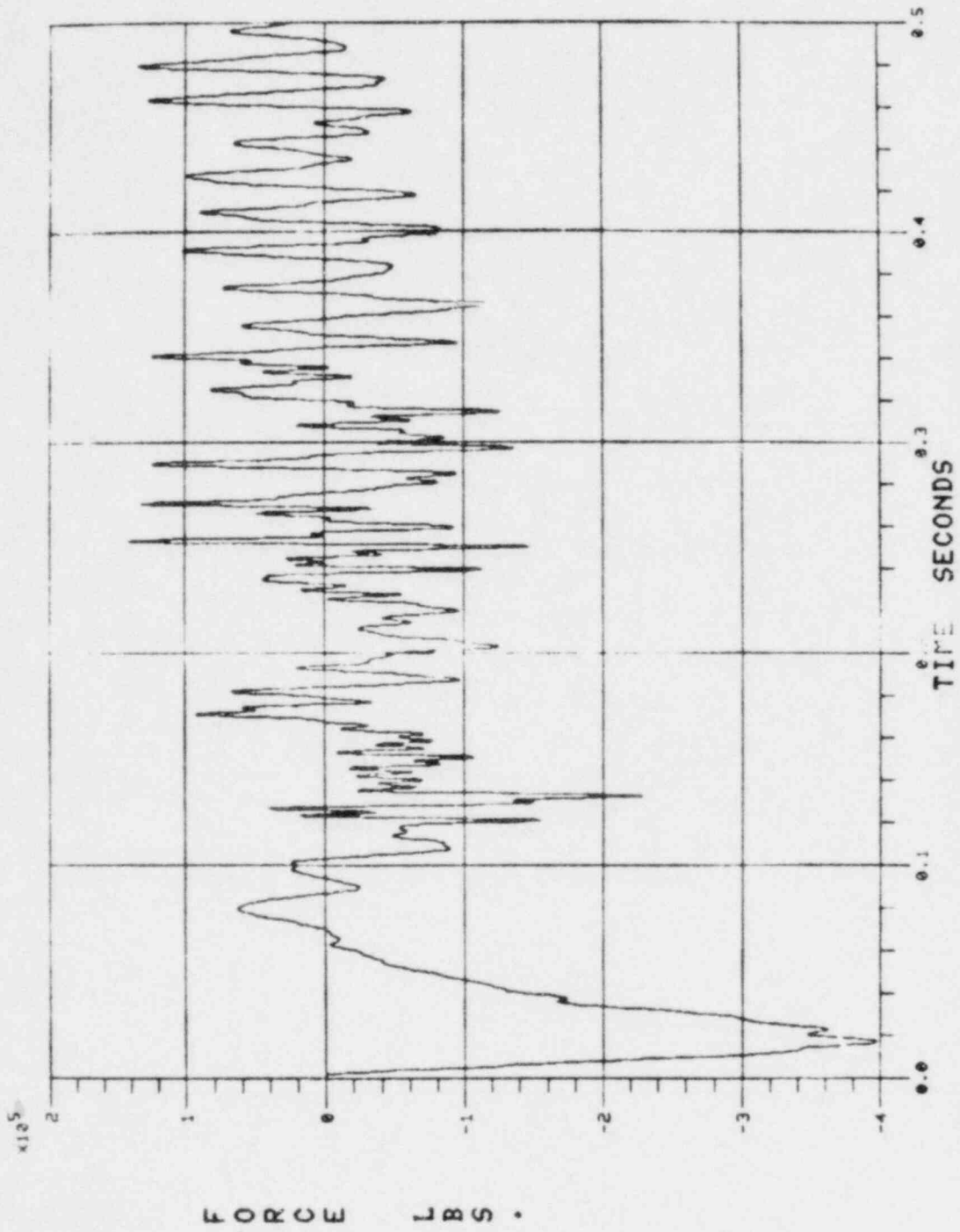
JT 44 FZ 45

Figure 2.21-1. (Cont'd)



JT 31 FX 49

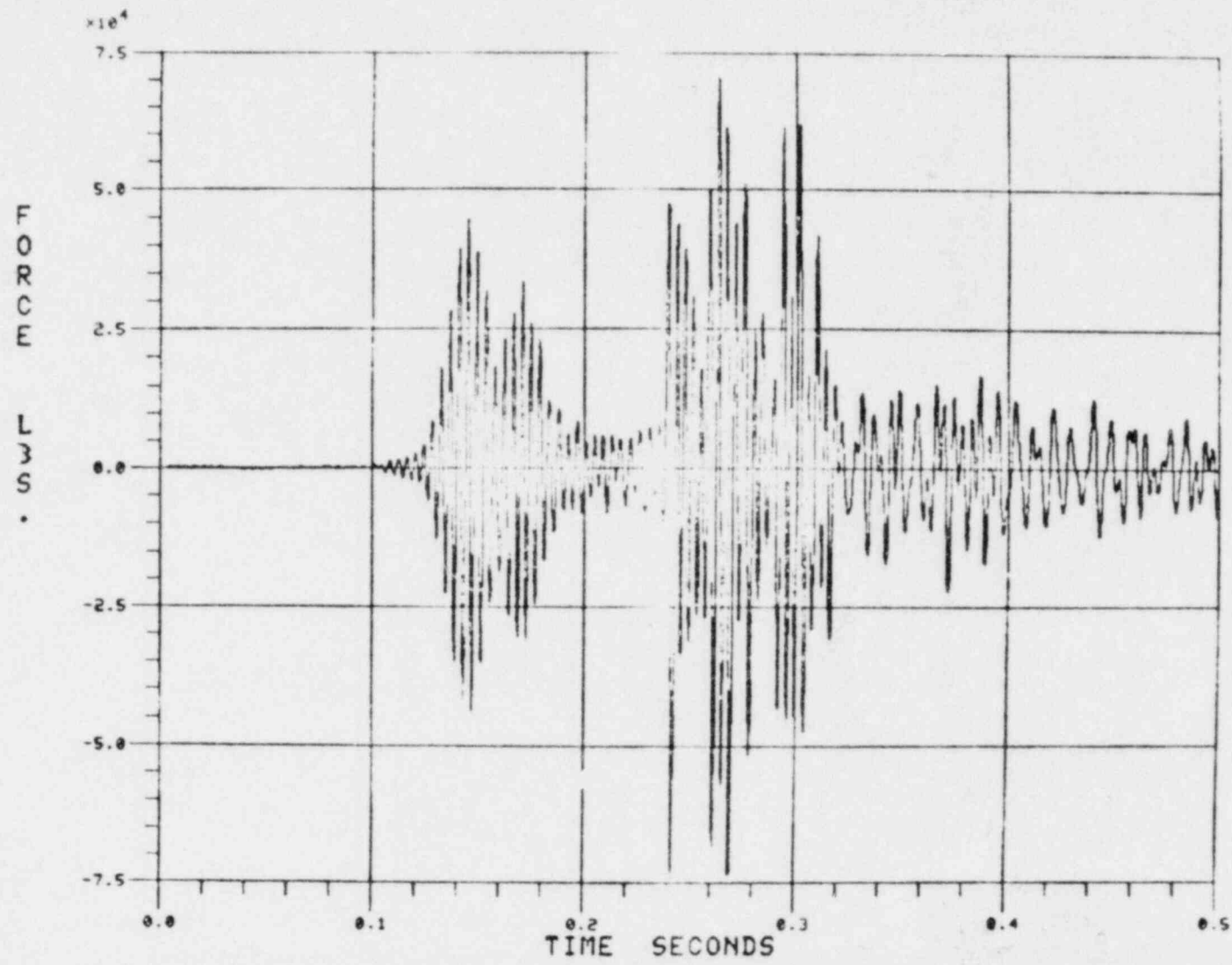
Figure 2.21-1. (Cont'd)



JT 31 FZ 51

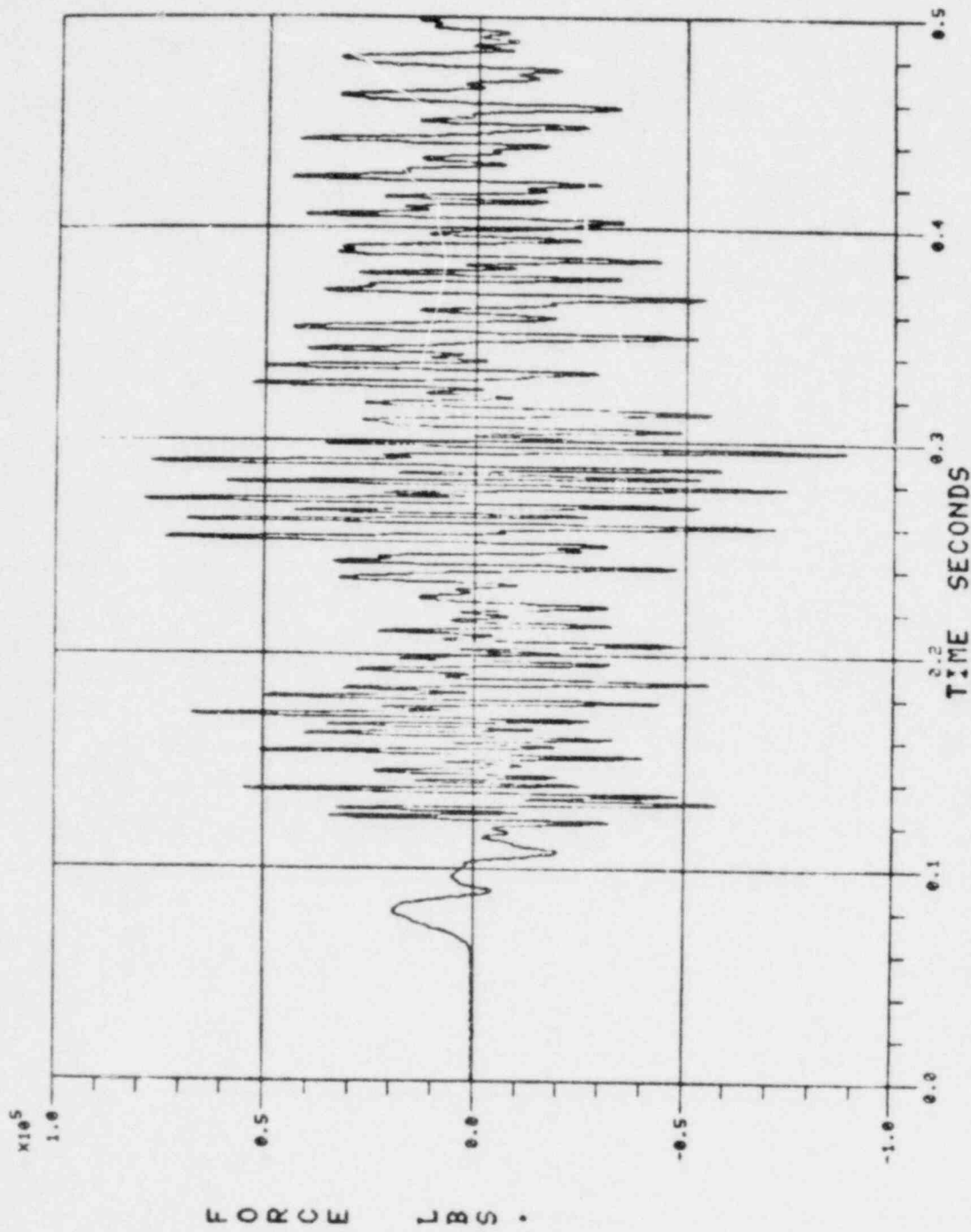


Figure 2.21-1. (Cont'd)



2.21-10

Figure 2.21-1. (Cont'd)



JT 7 FZ 87

Figure 2.21-1. (Cont'd)

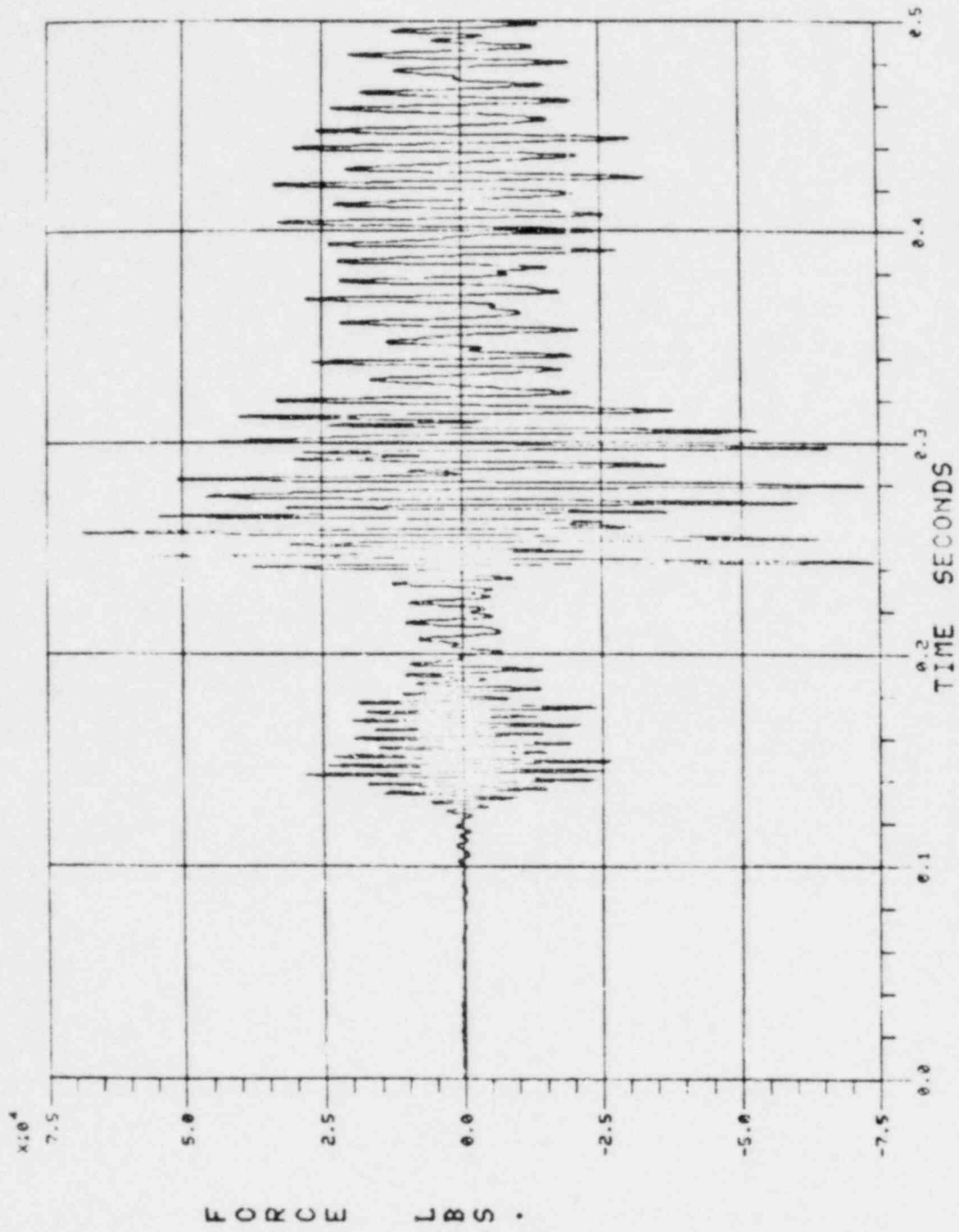
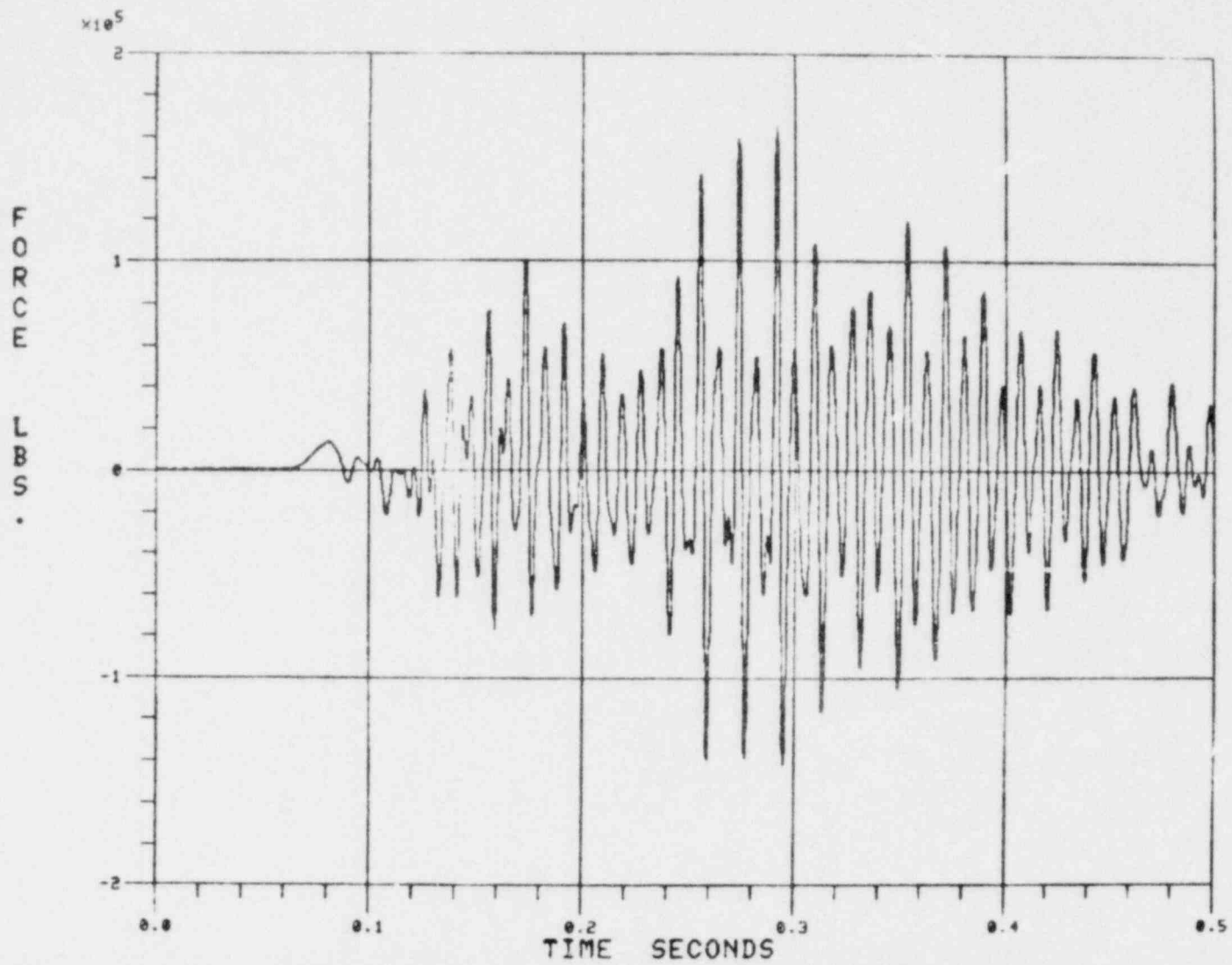
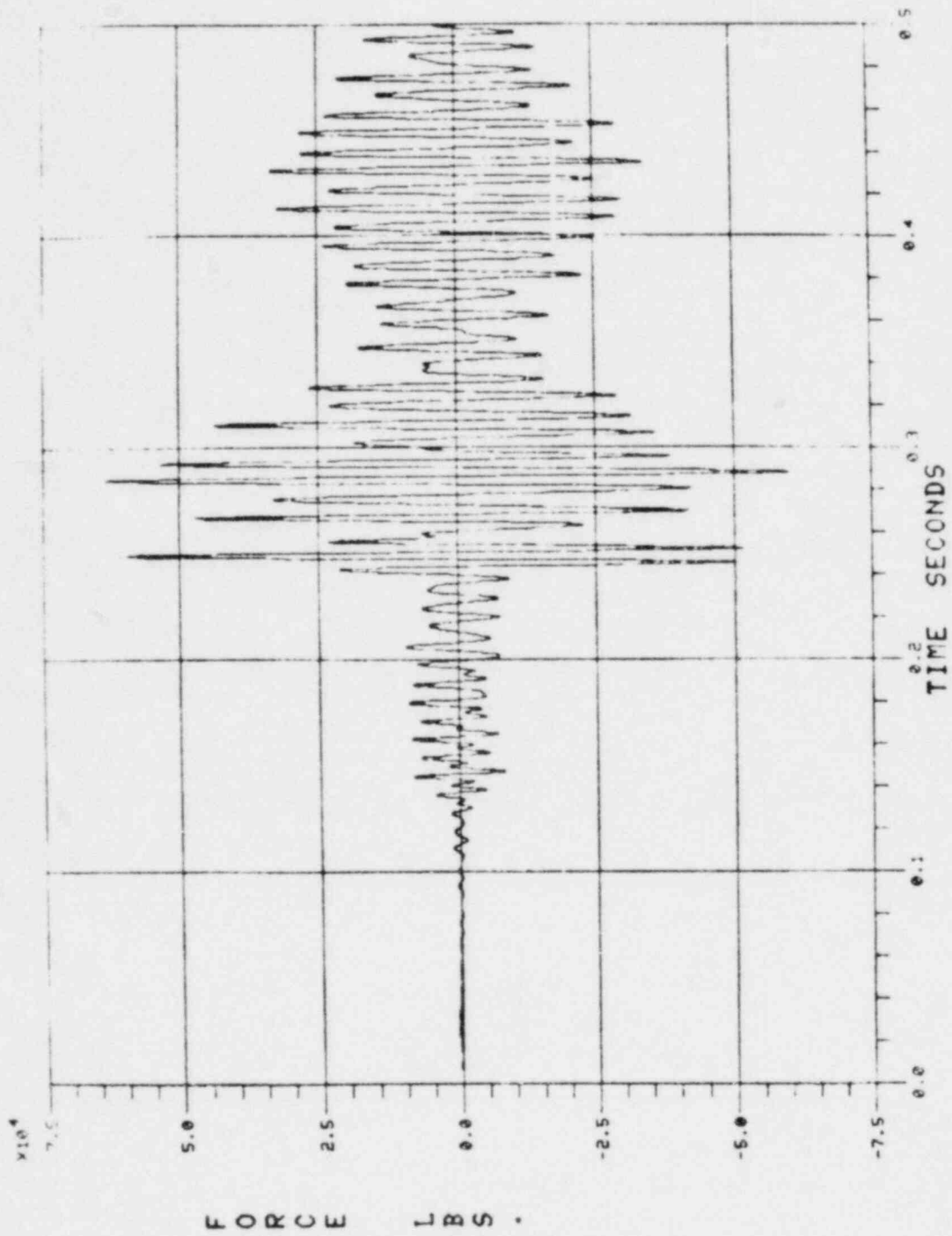


Figure 2.21-1. (Cont'd)



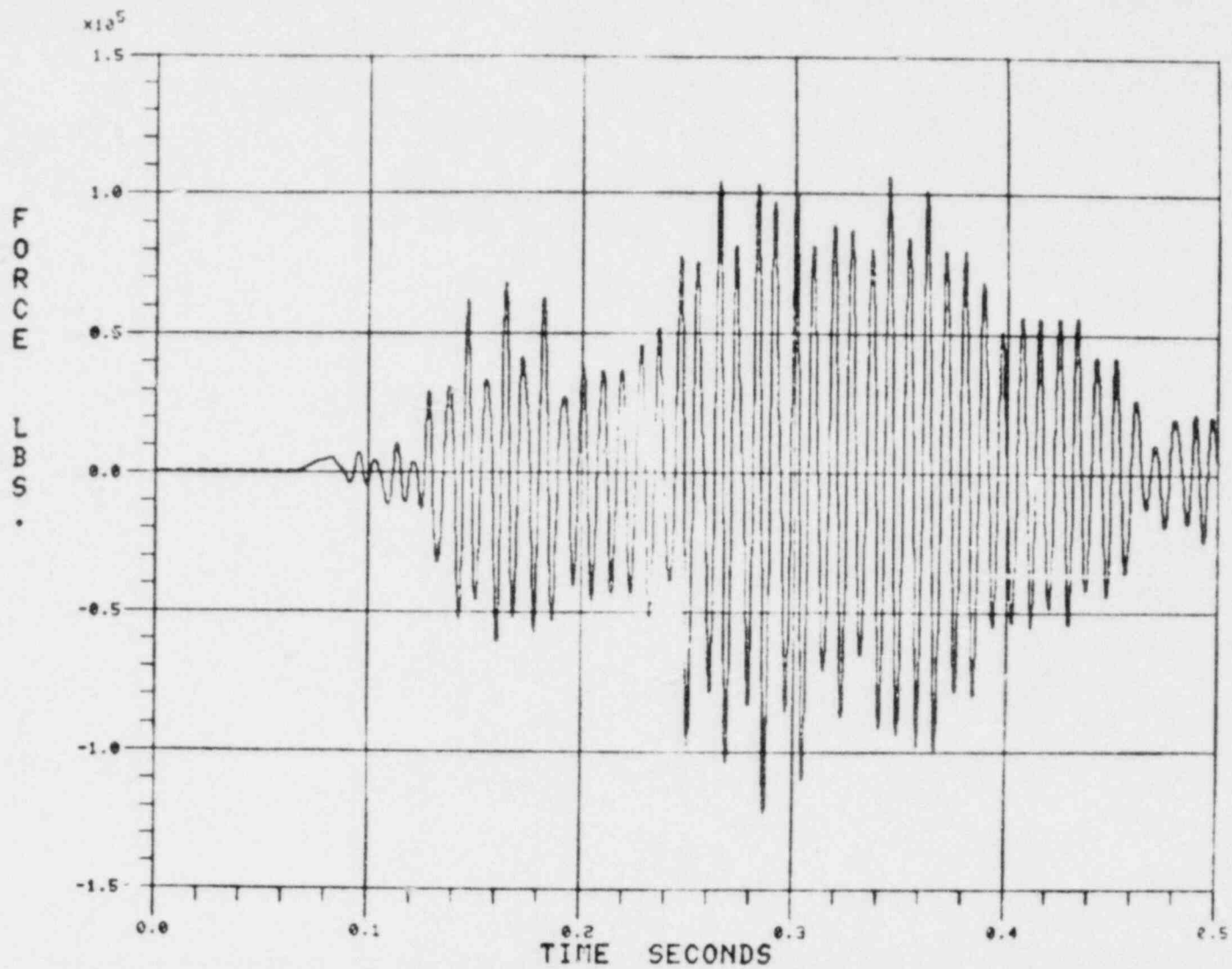
2.21-13

Figure 2.21-1. (Cont'd)



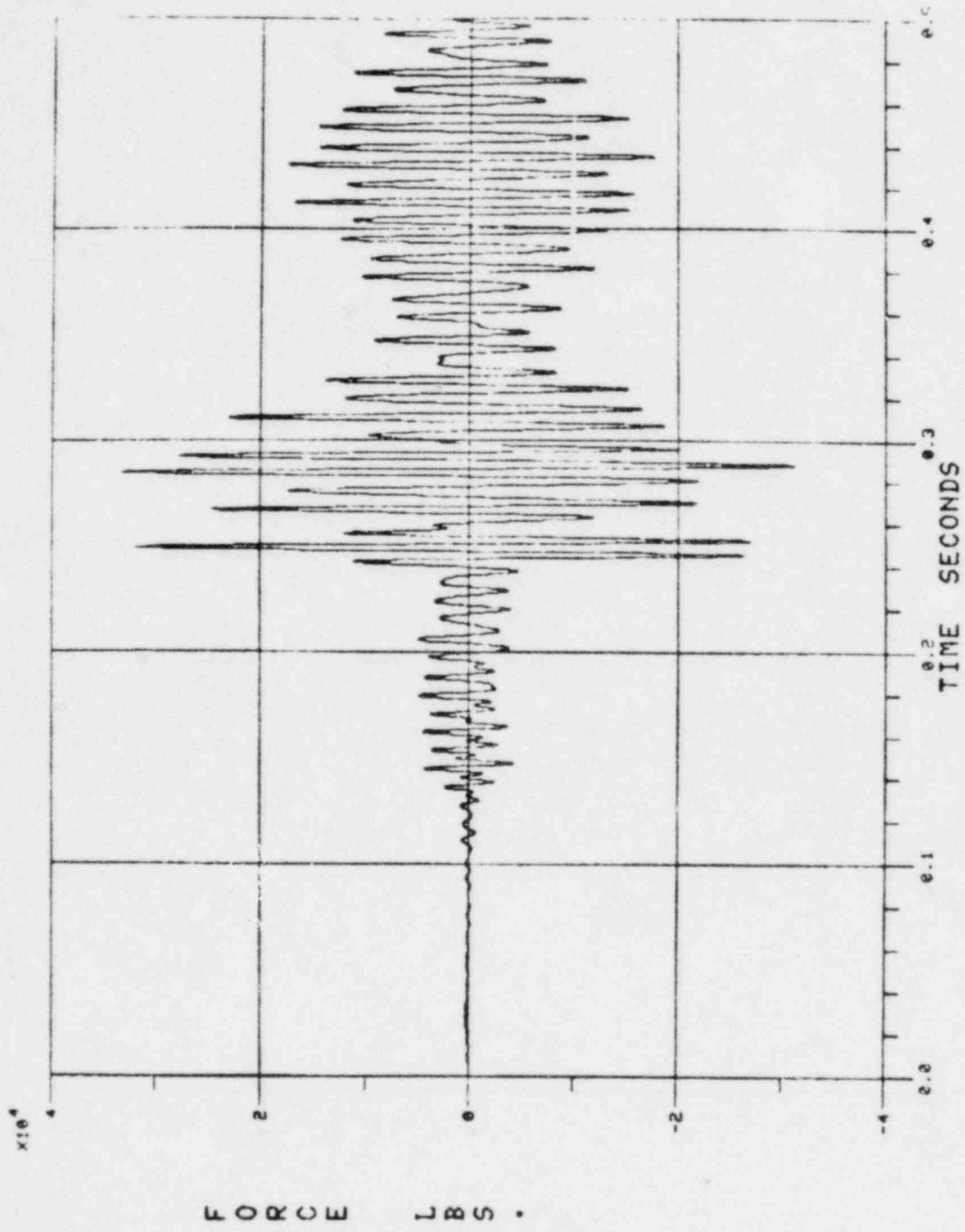
JT 6 FX 97

Figure 2.21-1. (Cont'd)



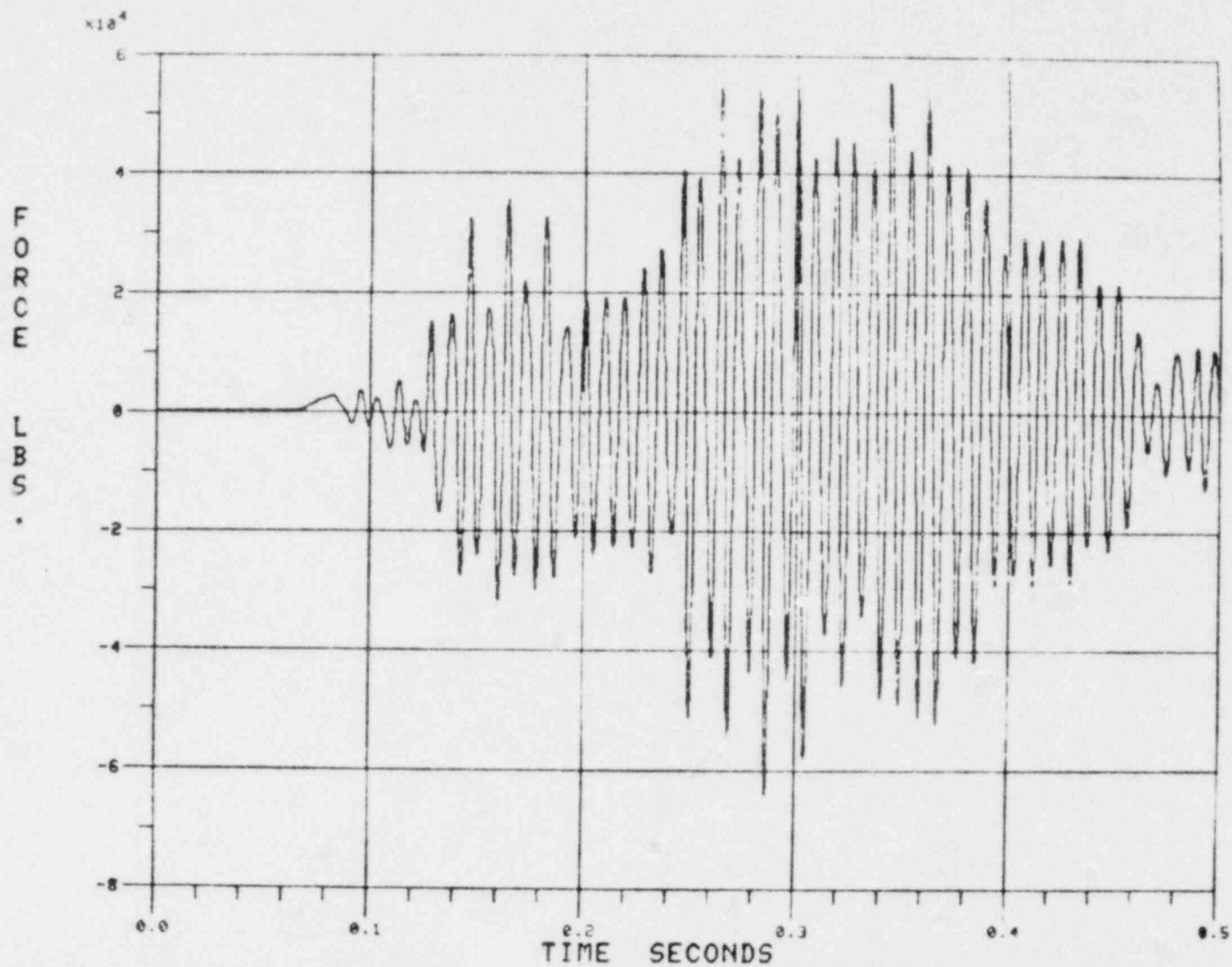
2.21-15

Figure 2.21-1. (Cont'd)



JT 10 FX 103

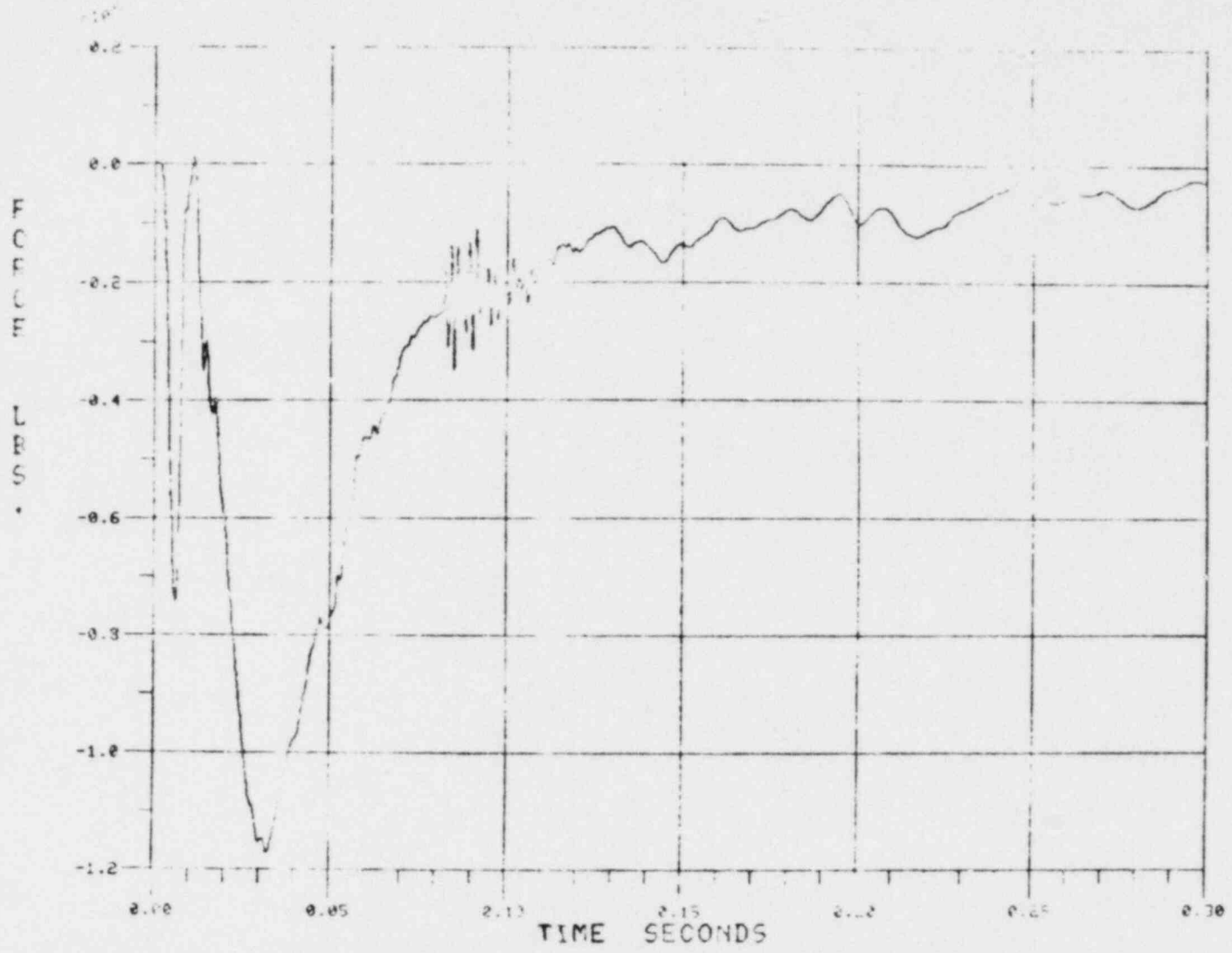
Figure 2.21-1. (Cont'd)



2.21-17

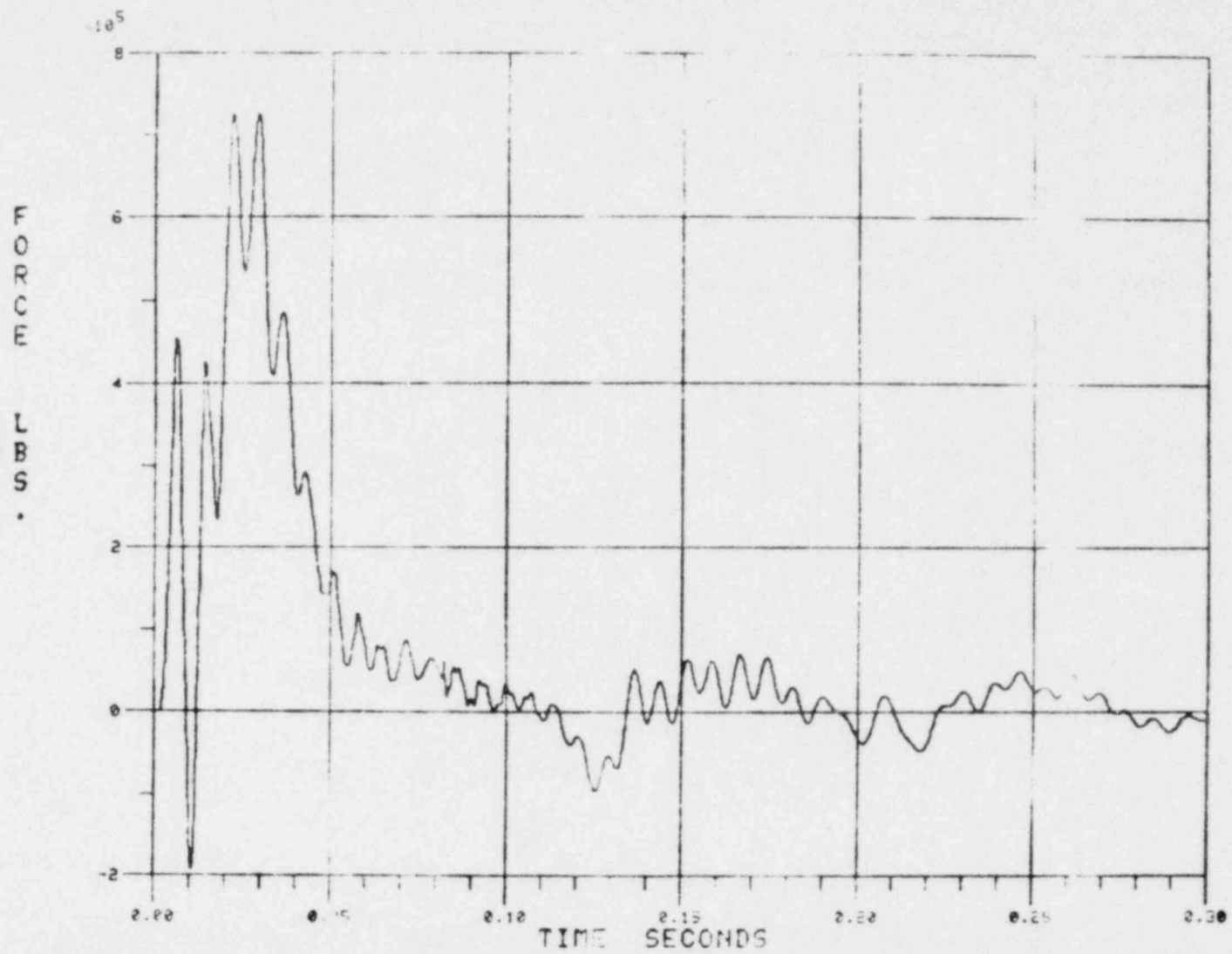


Figure 2.21-2. Force Across Core Support Shield 2A Cold Leg Break at RV-Skirt-Supported Plant



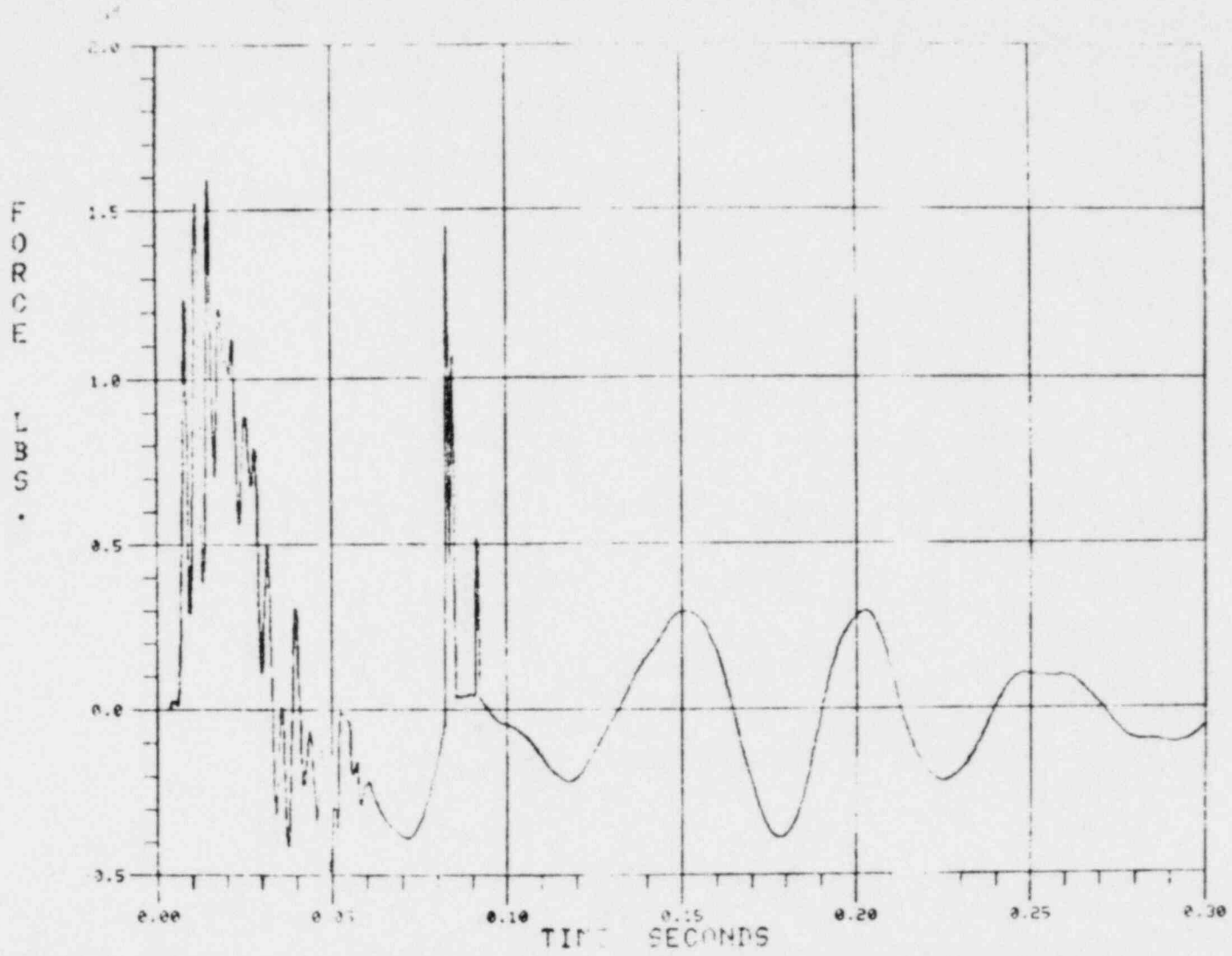
2.21-18

Figure 2.21-2. (Cont'd)



2.21-19

Figure 2.21-2. (Cont'd)



2.21-20

Figure 2.21-2. (Cont'd)

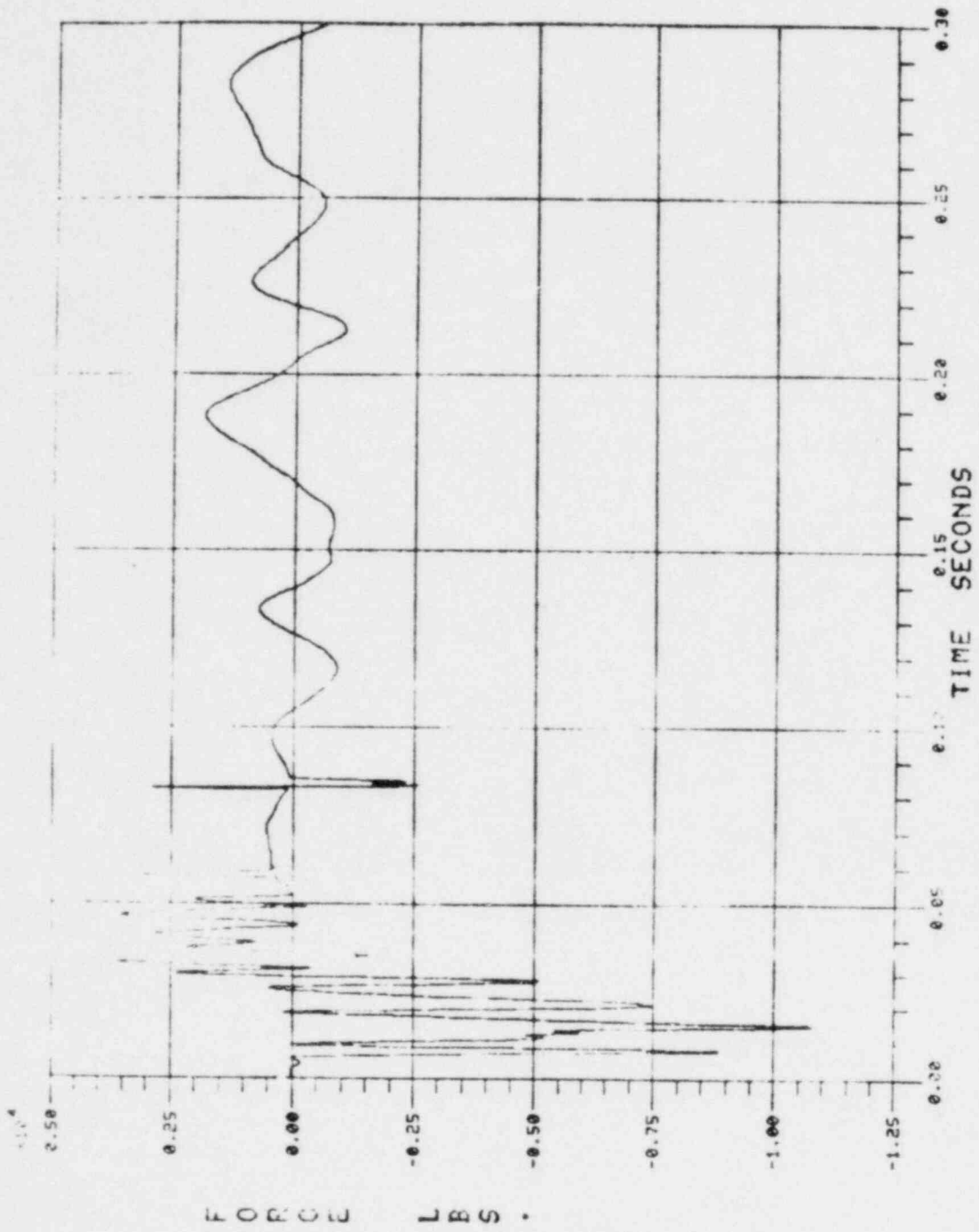
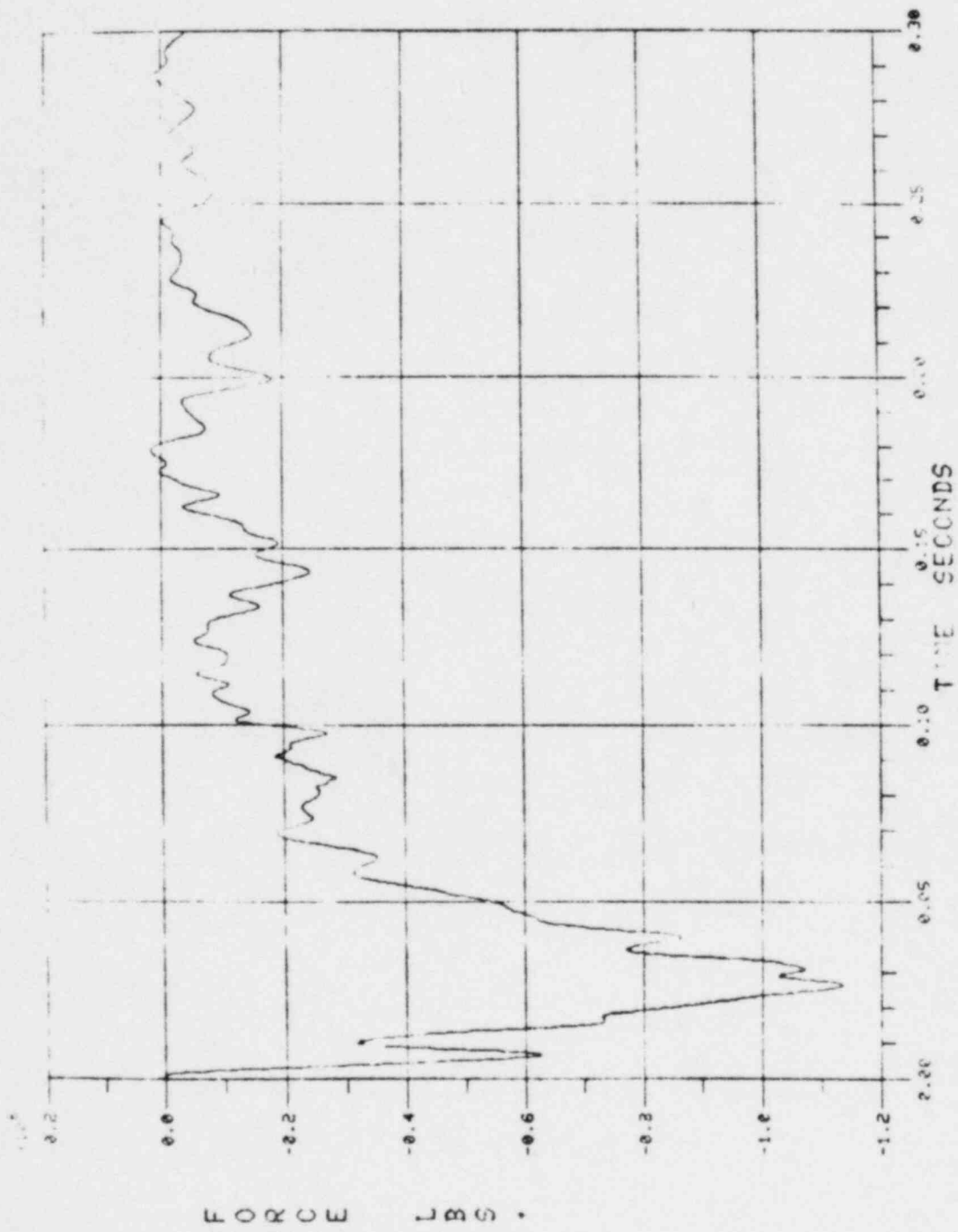
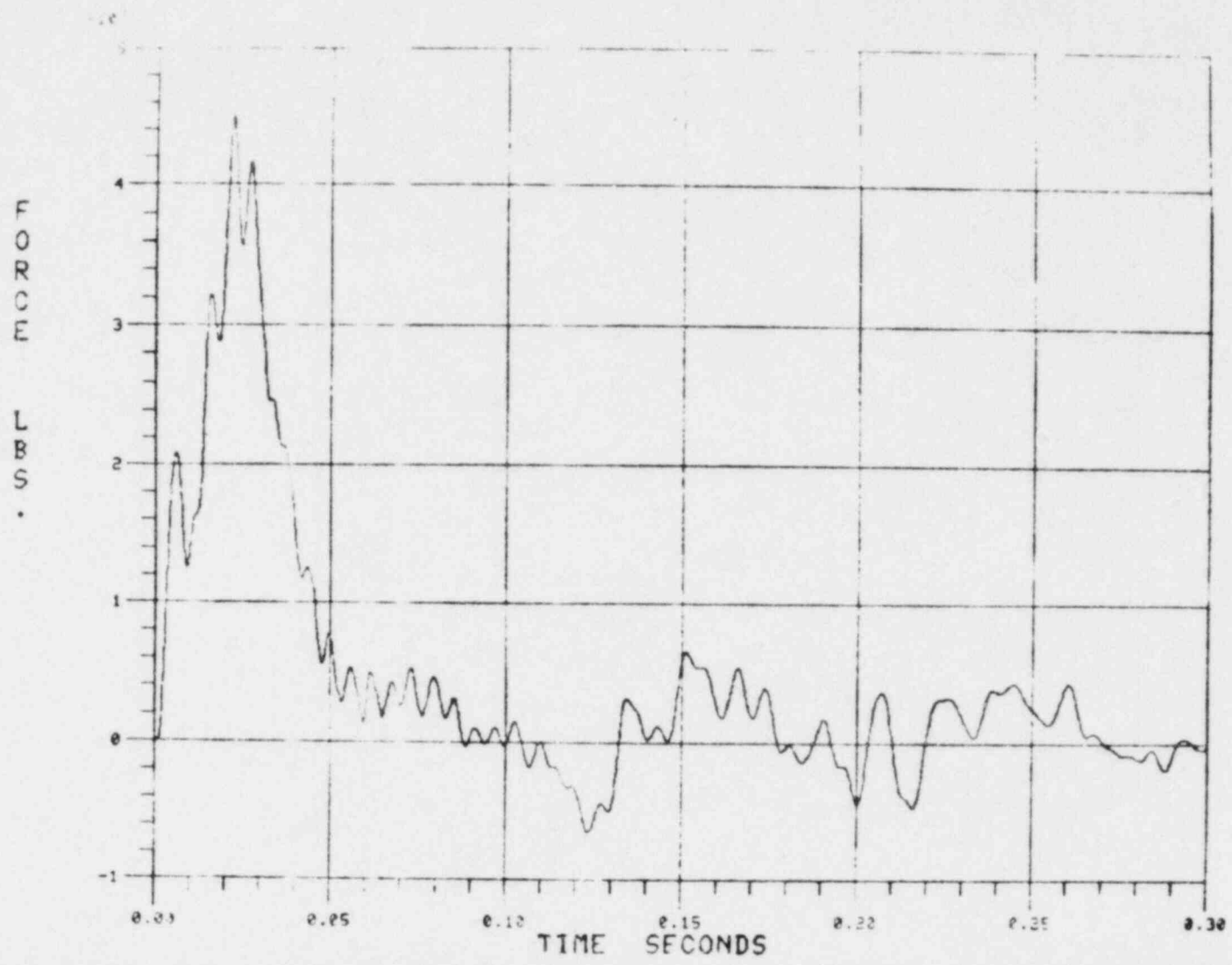


Figure 2.21-2. (Cont'd)



JT 44 FX 43

Figure 2.21-2. (Cont'd)



2.21-23

Figure 2.21-2. (Cont'd)

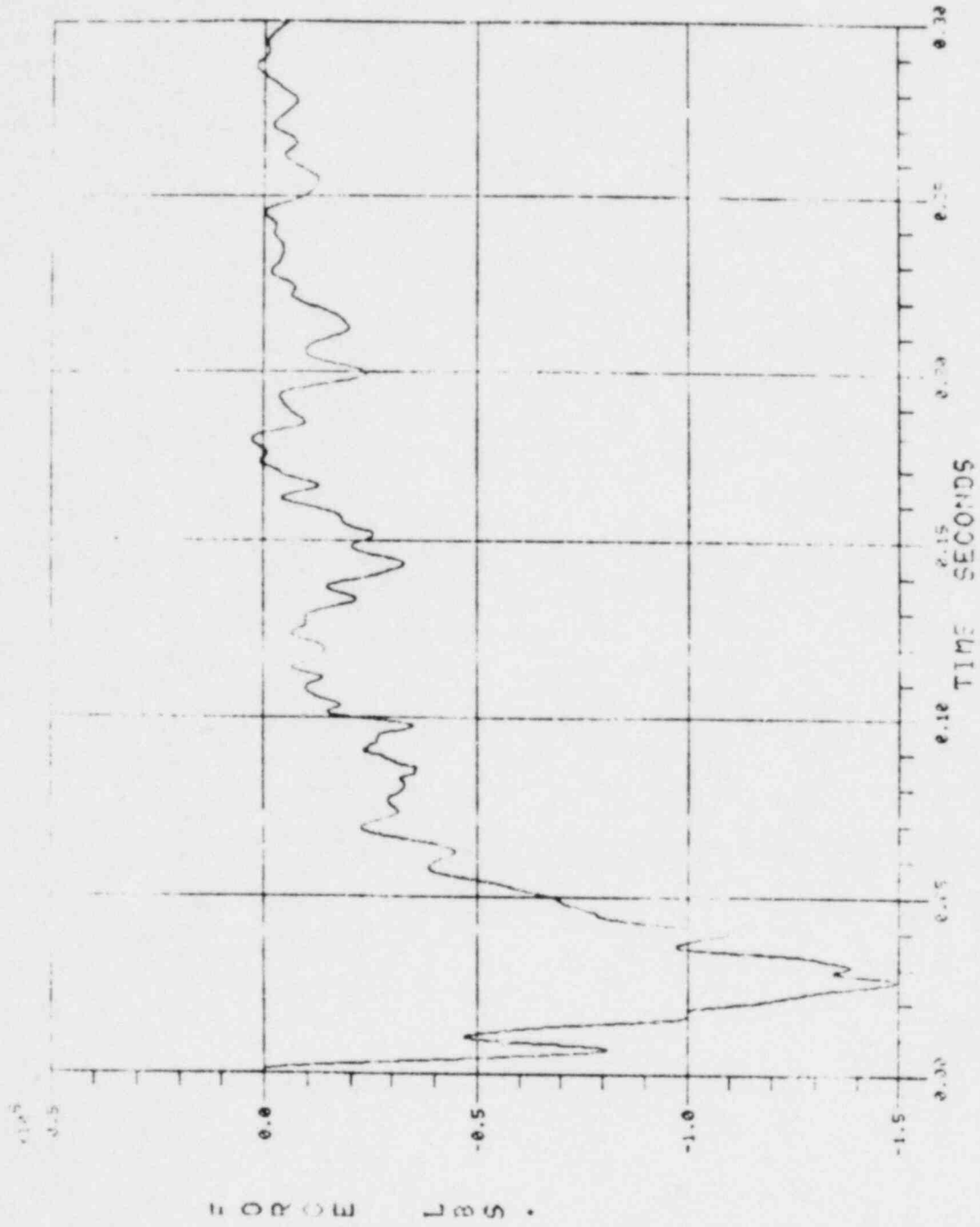
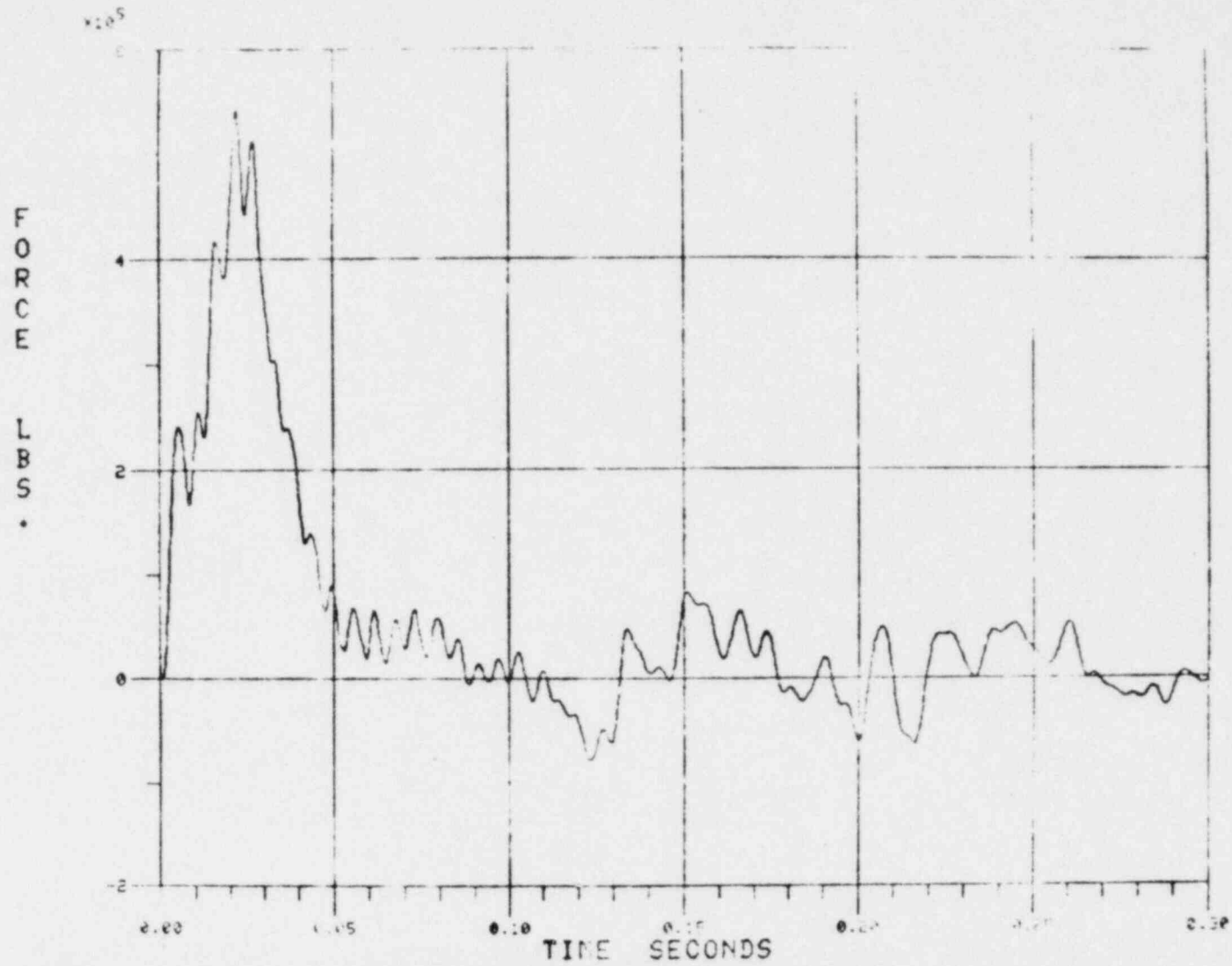


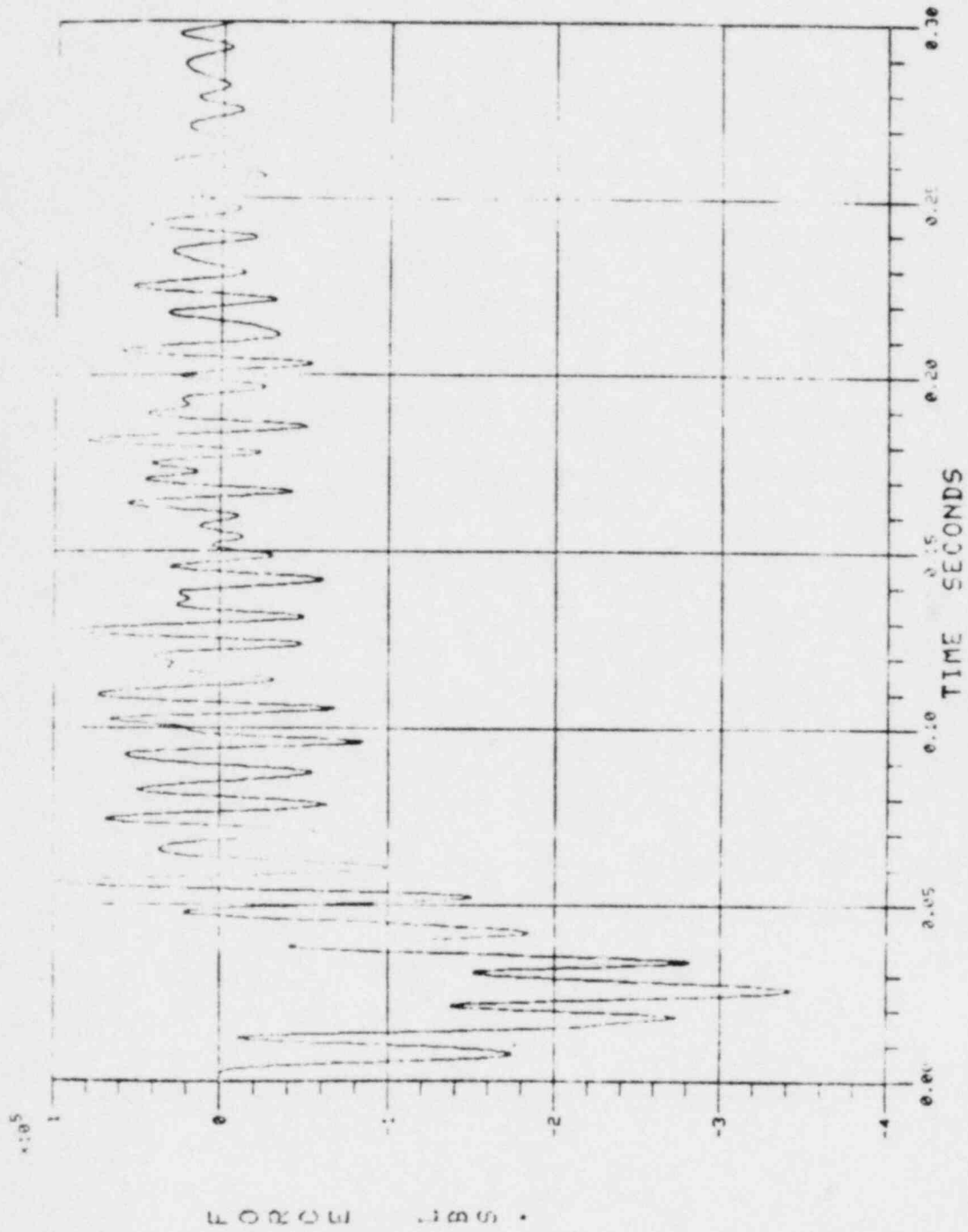
Figure 2.21-2. (Cont'd)



2.21-25



Figure 2.21-2. (Cont'd)



JT 1 FX 91

Figure 2.21-2. (Cont'd)

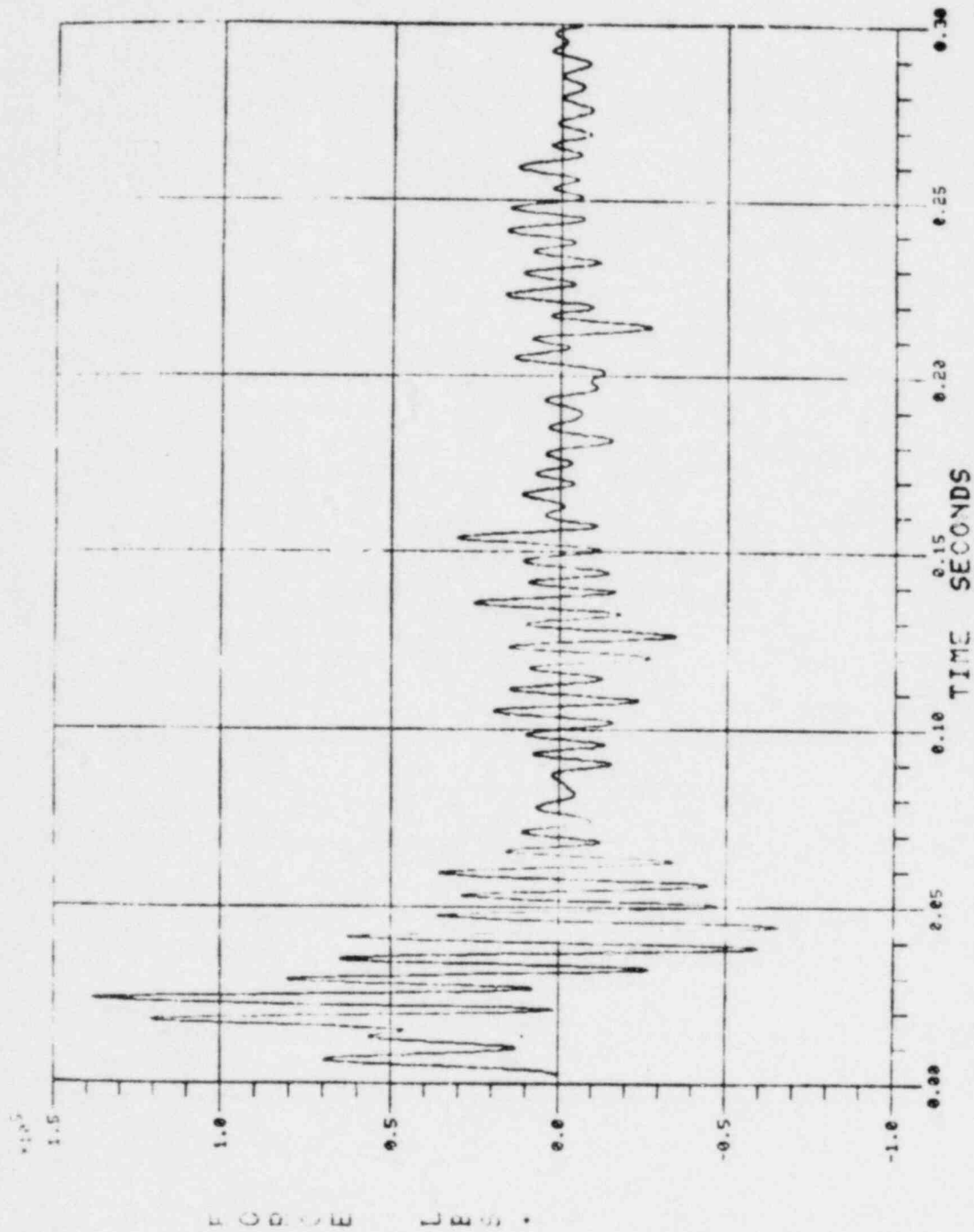
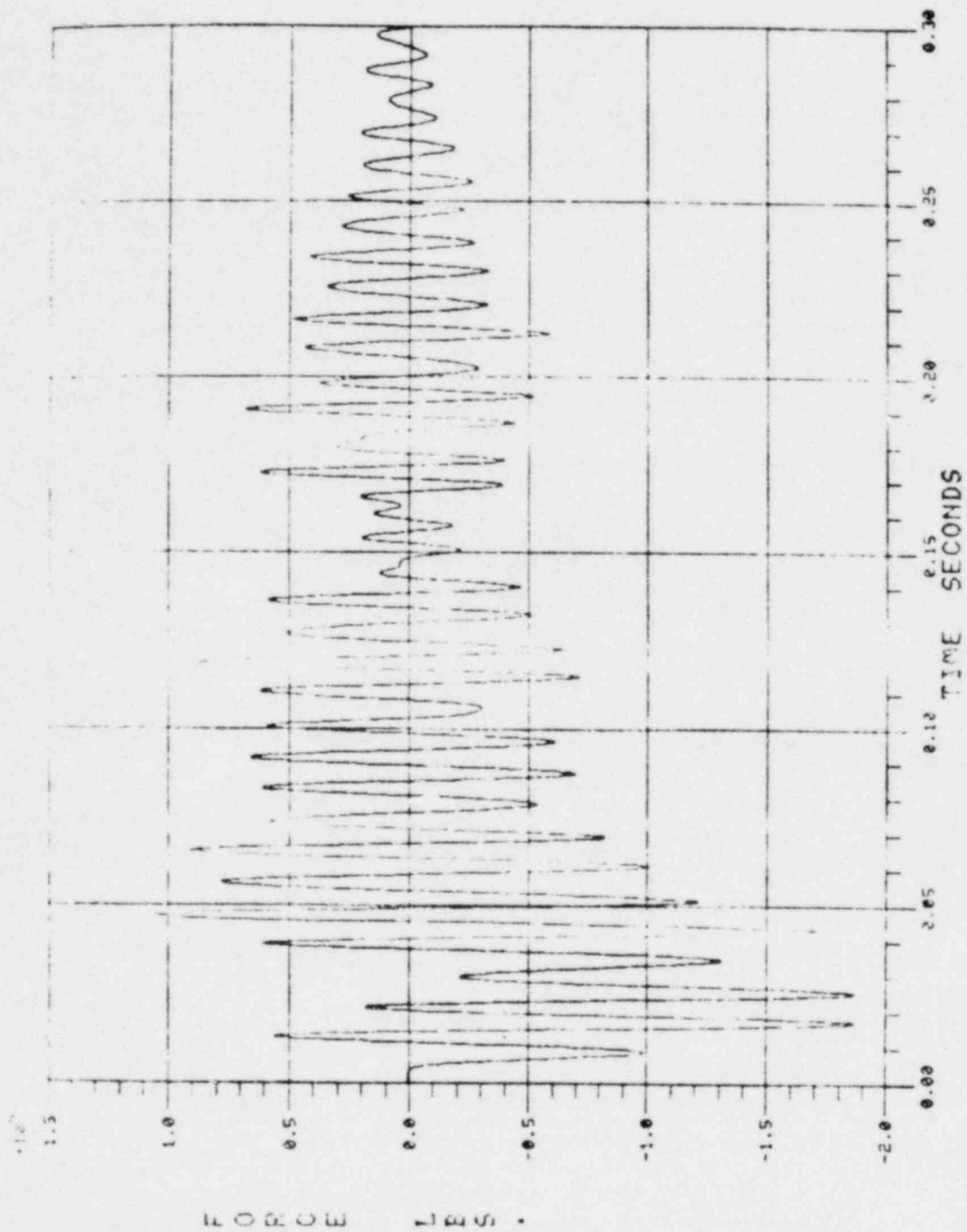
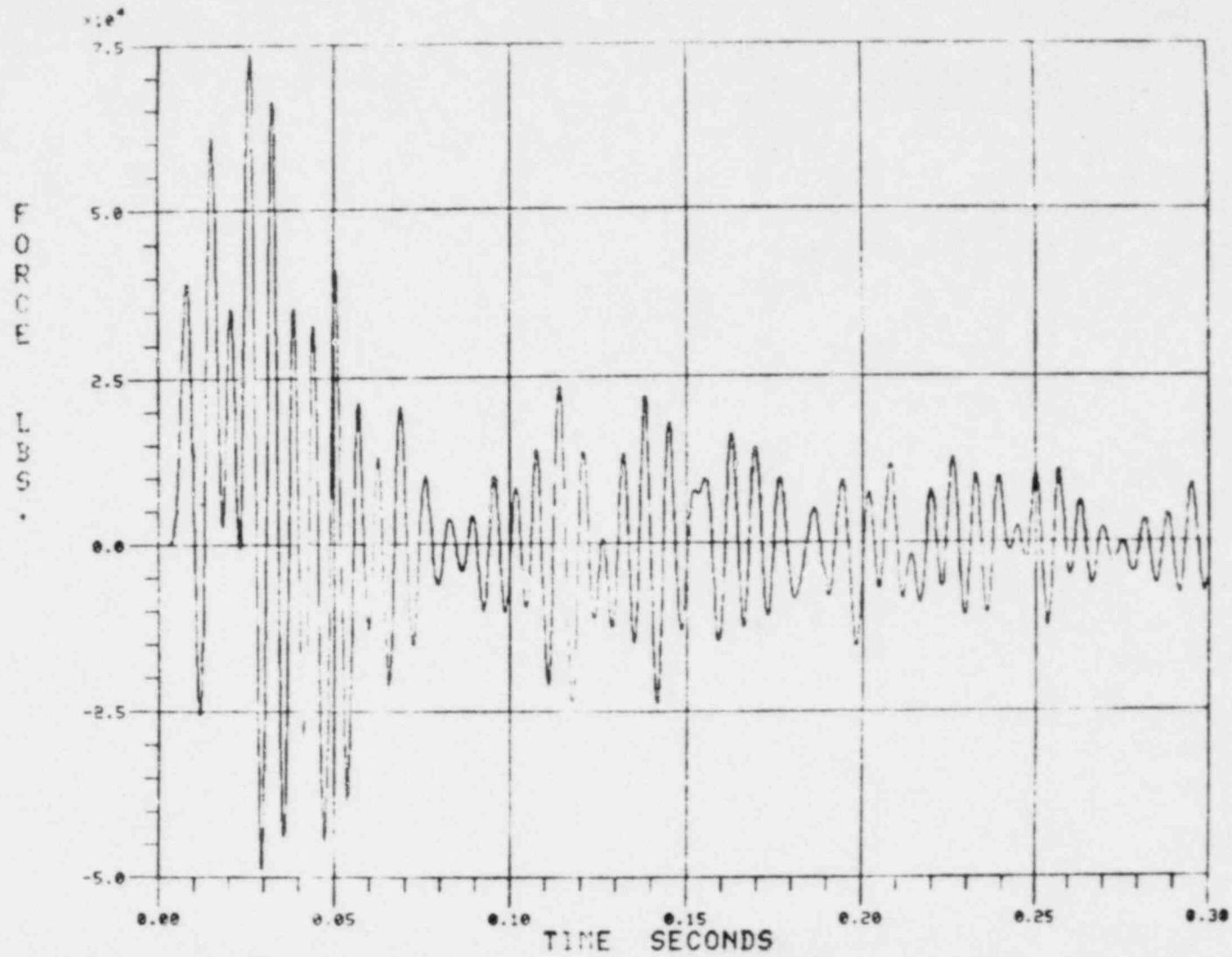


Figure 2.21-2. (Cont'd)



JT 6 FX 97

Figure 2.21-2. (Cont'd)



2.21-29

Figure 2.21-2. (Cont'd)

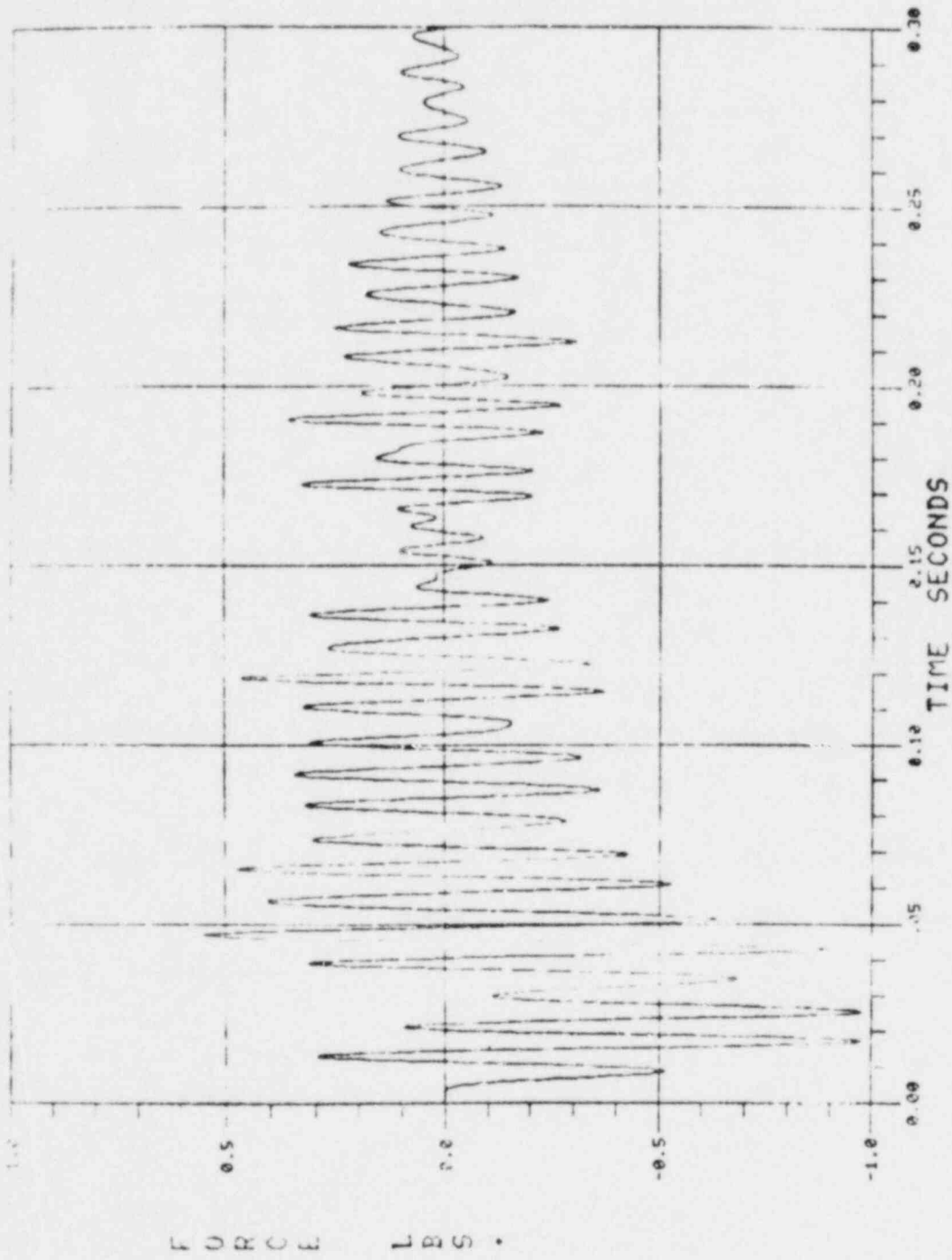
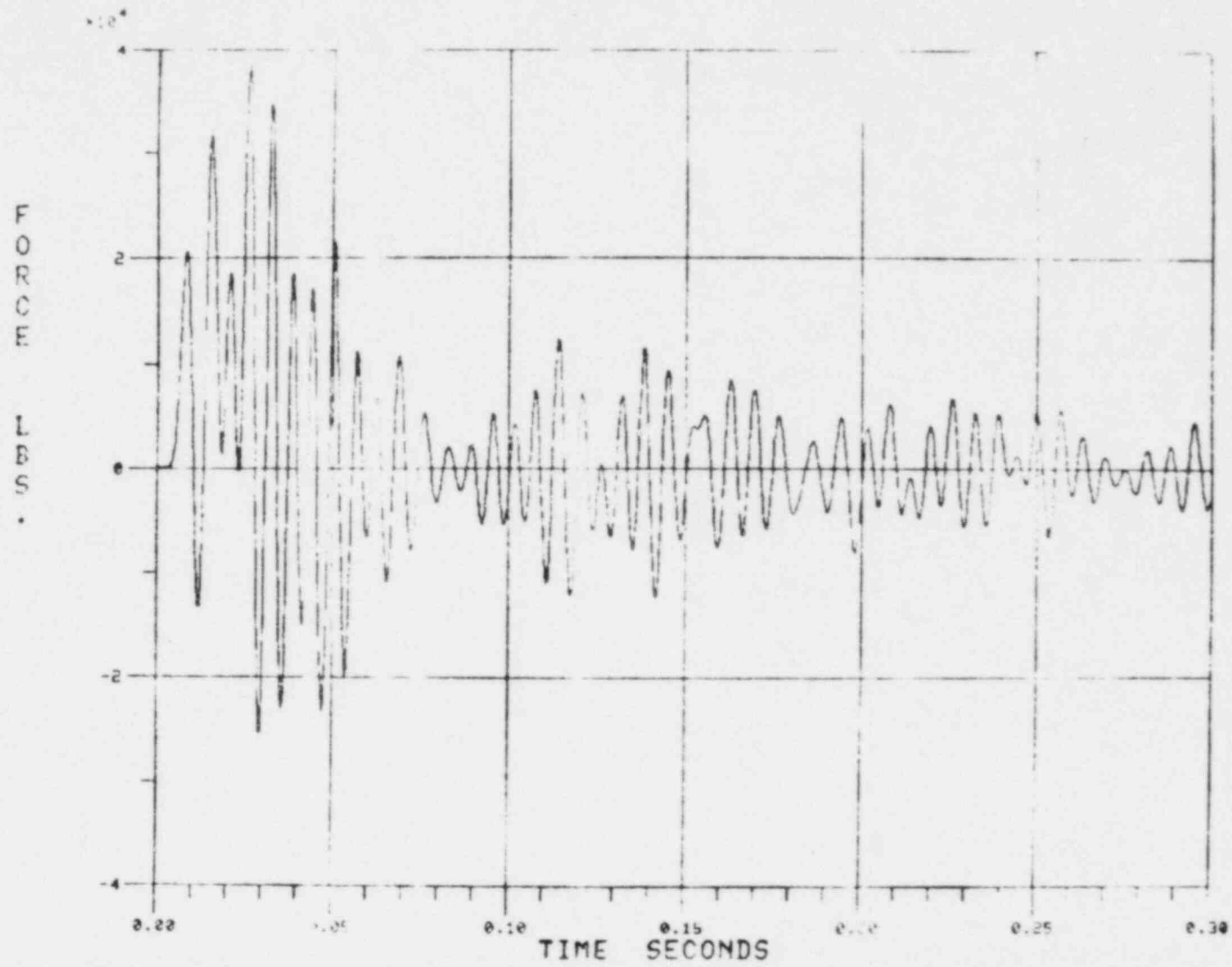
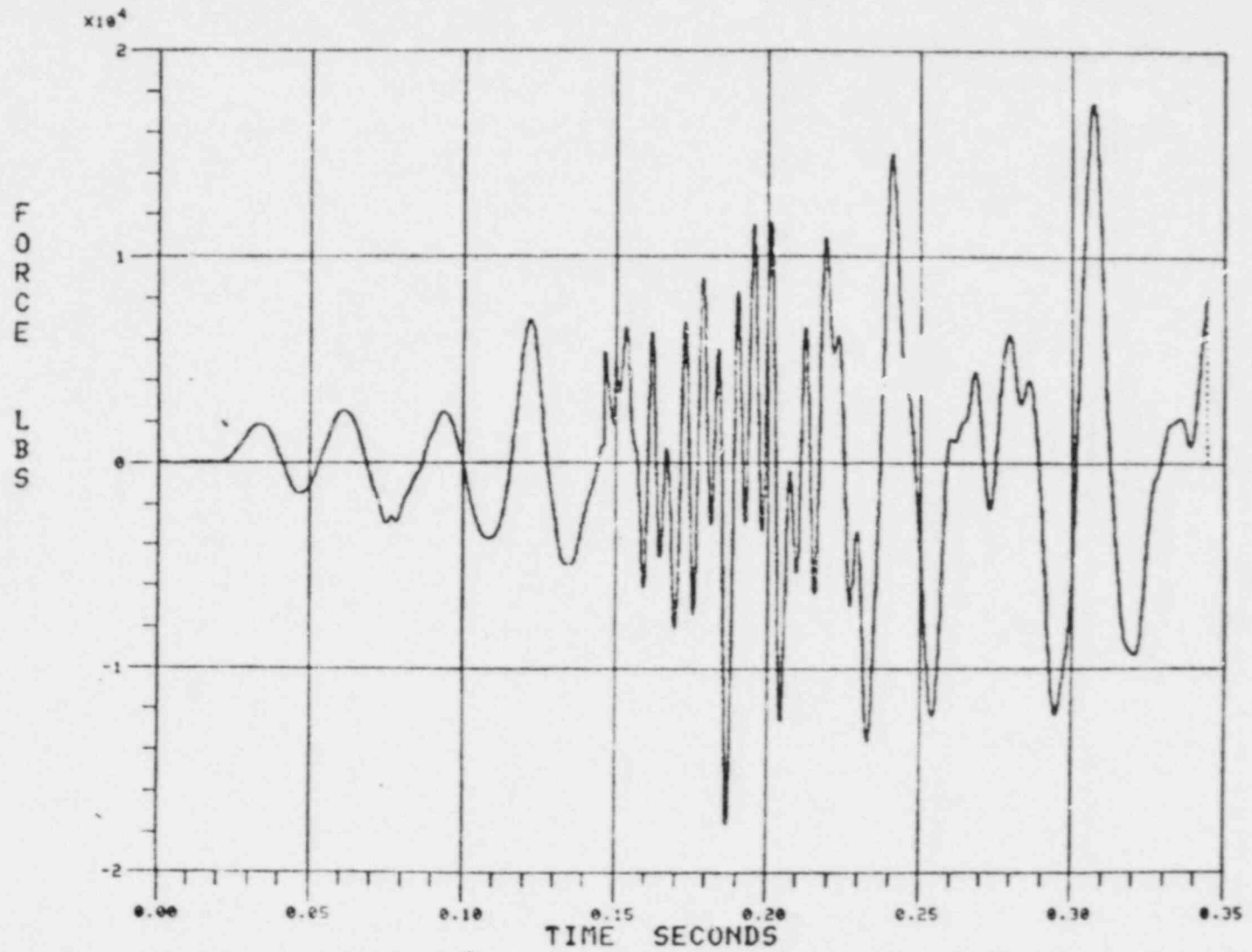


Figure 2.21-2. (Cont'd)



2.21-31

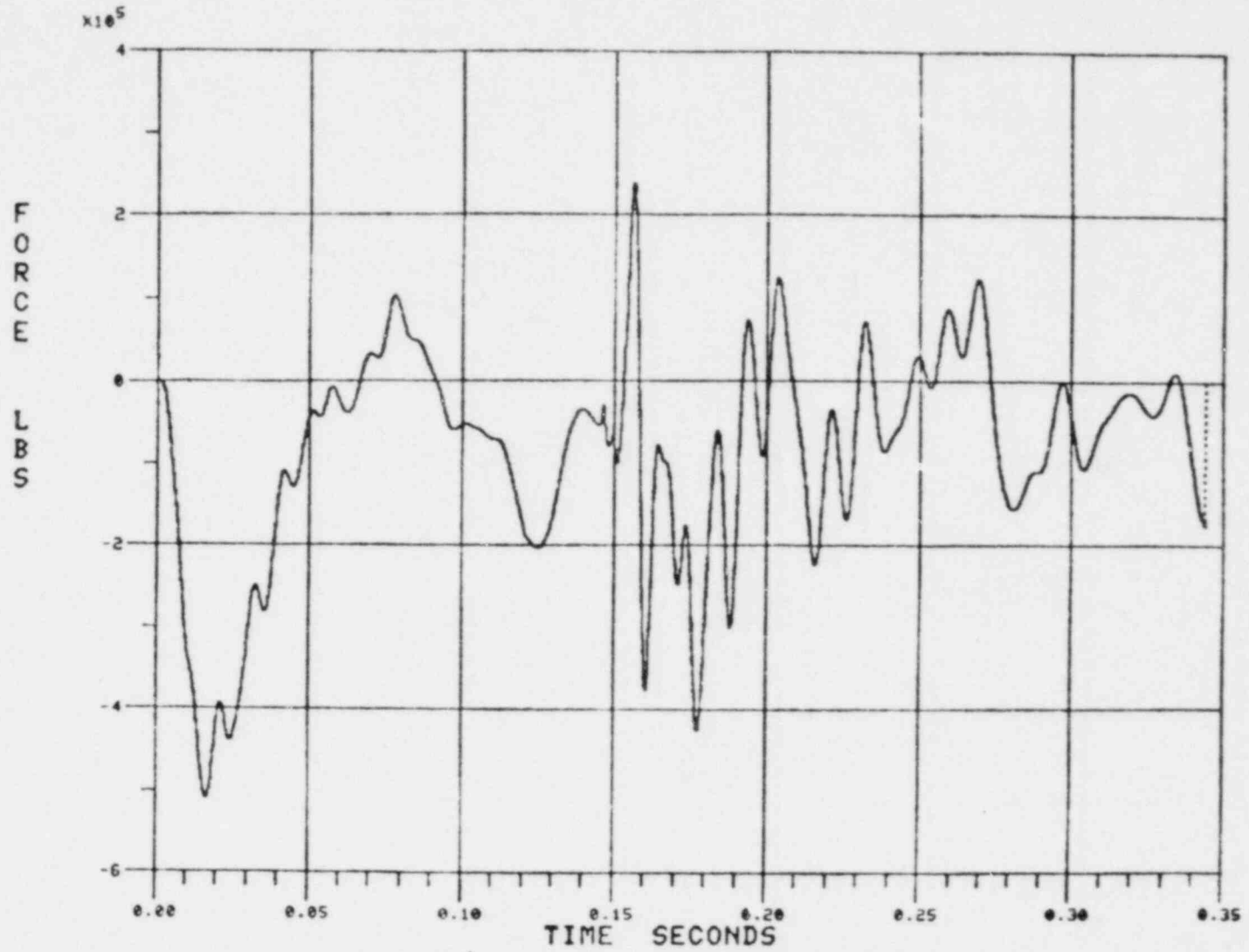
Figure 2.21-3. Force Across Core Support Shield - 1.024A Hot Leg Break at RV. Nozzle-Supported Plant



JOINT 31 X DIR CORE SUPPORT SHEILD

2.21-32

Figure 2.21-3. (Cont'd)

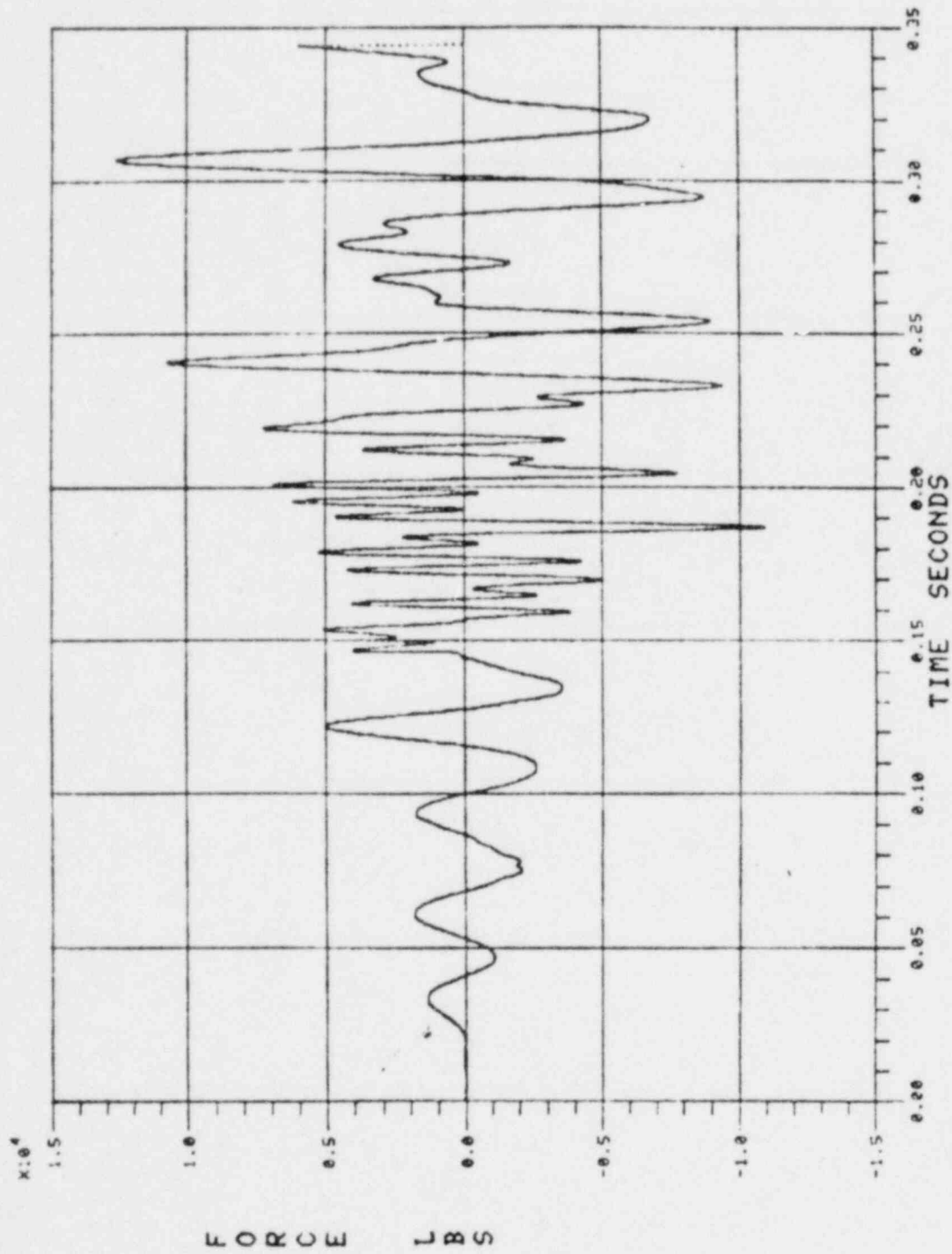


2.21-33

JOINT 31 Z DIR CORE SUPPORT SHEILD

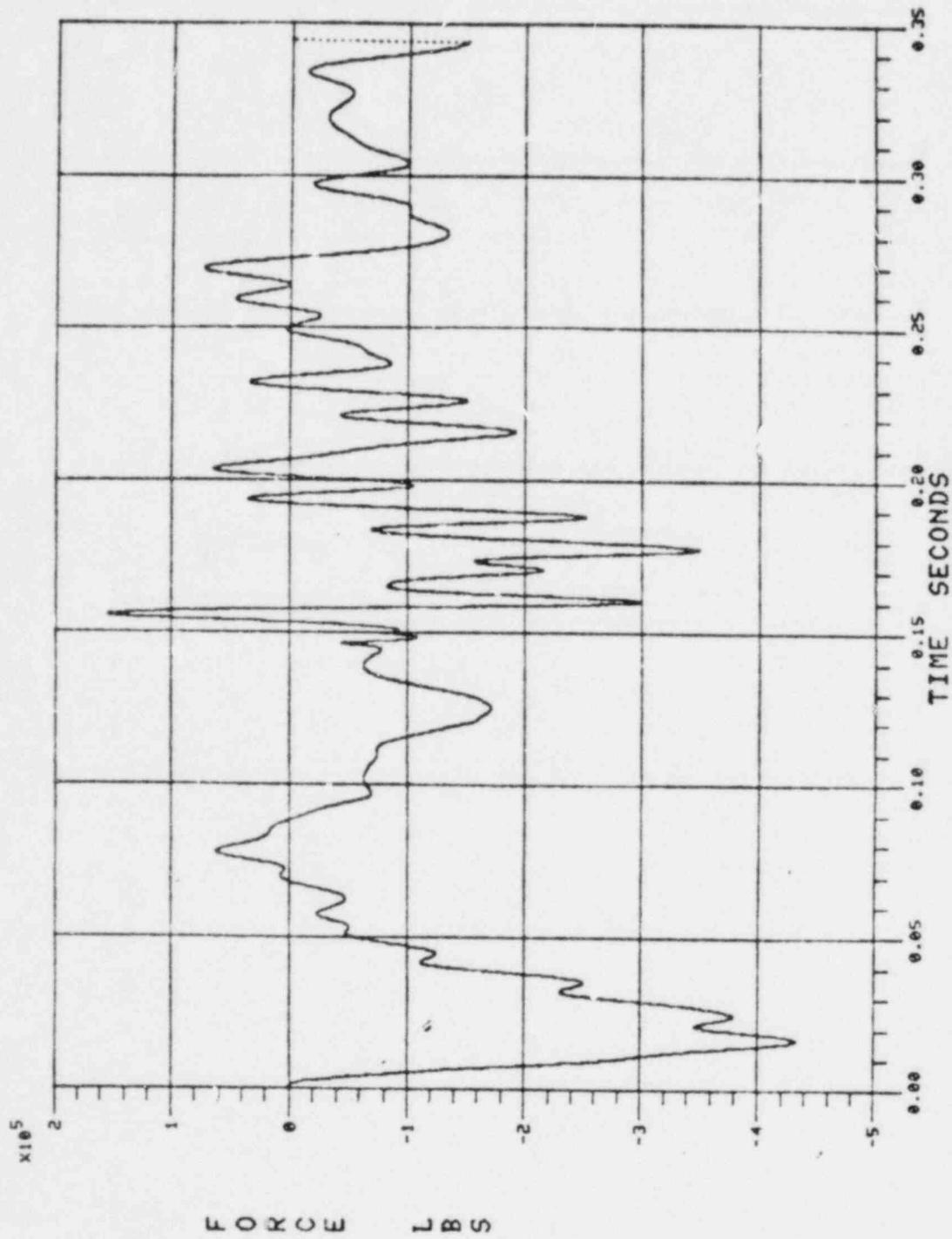


Figure 2.21-3. (Cont'd)



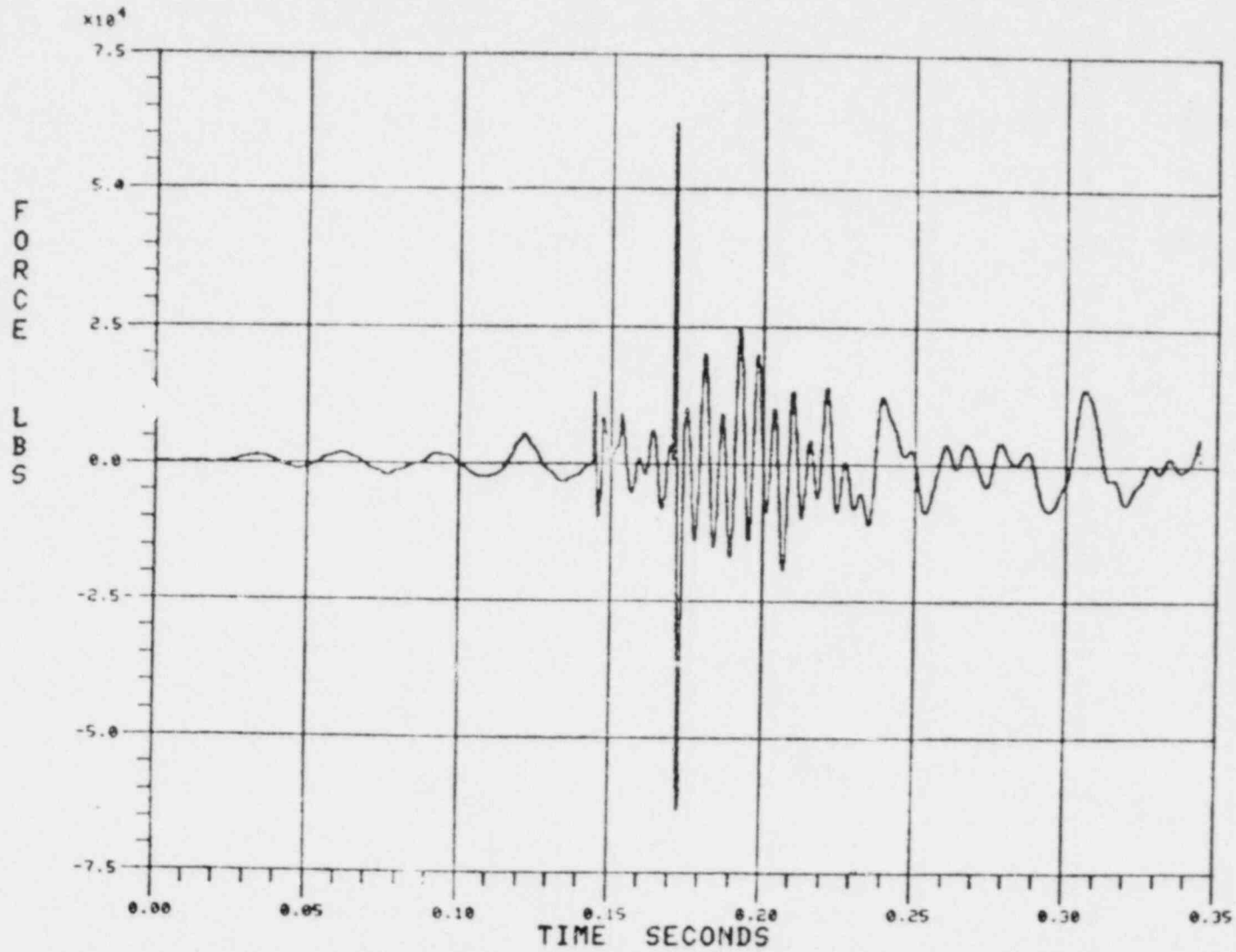
JOINT 44 X DIR CORE SUPPORT SHIELD

Figure 2.21-3. (Cont'd)



JOINT 44 Z DIR CORE SUPPORT SHIELD

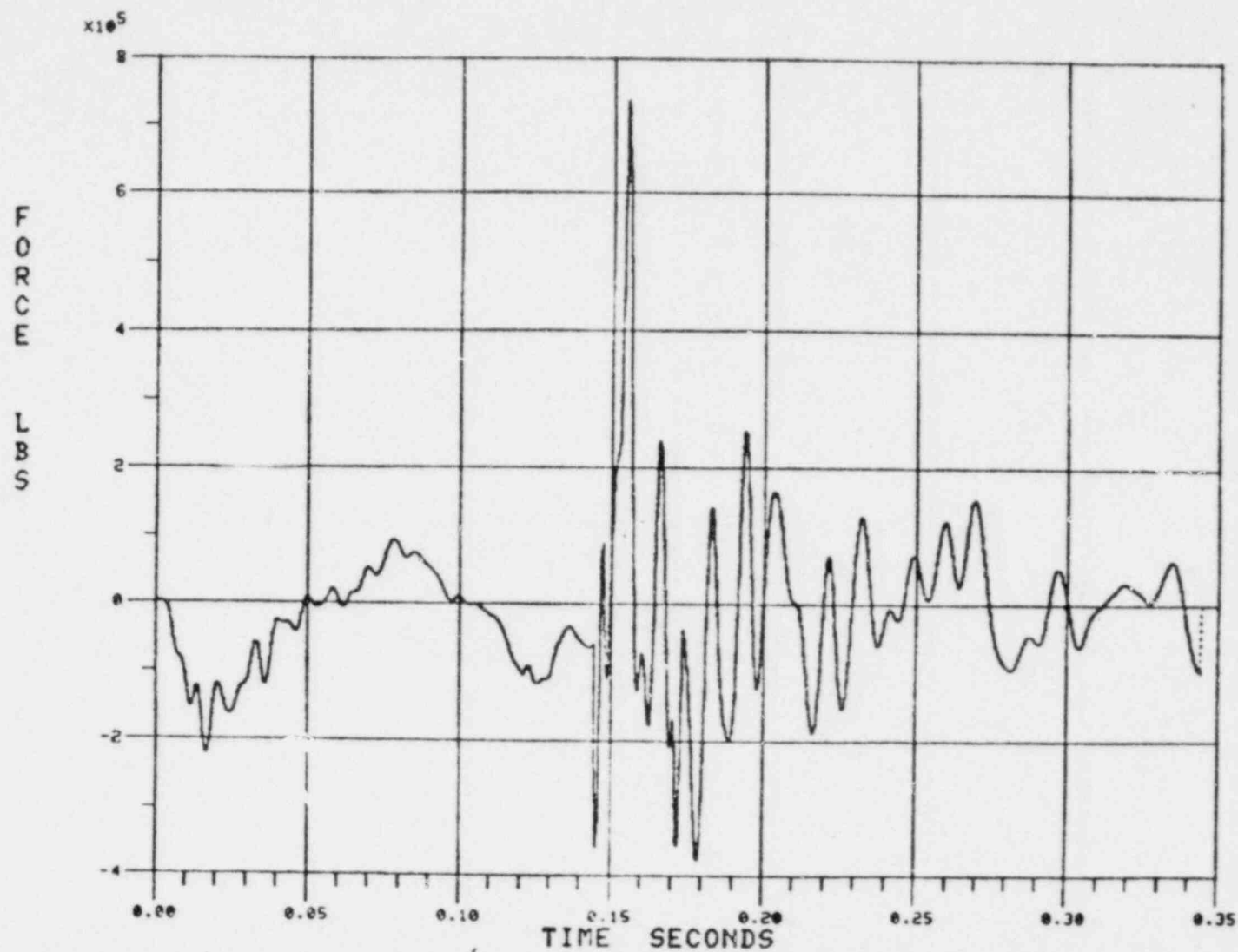
Figure 2.21-3. (Cont'd)



2.21-36

JOINT 62 X DIR CORE SUPPORT SHEILD

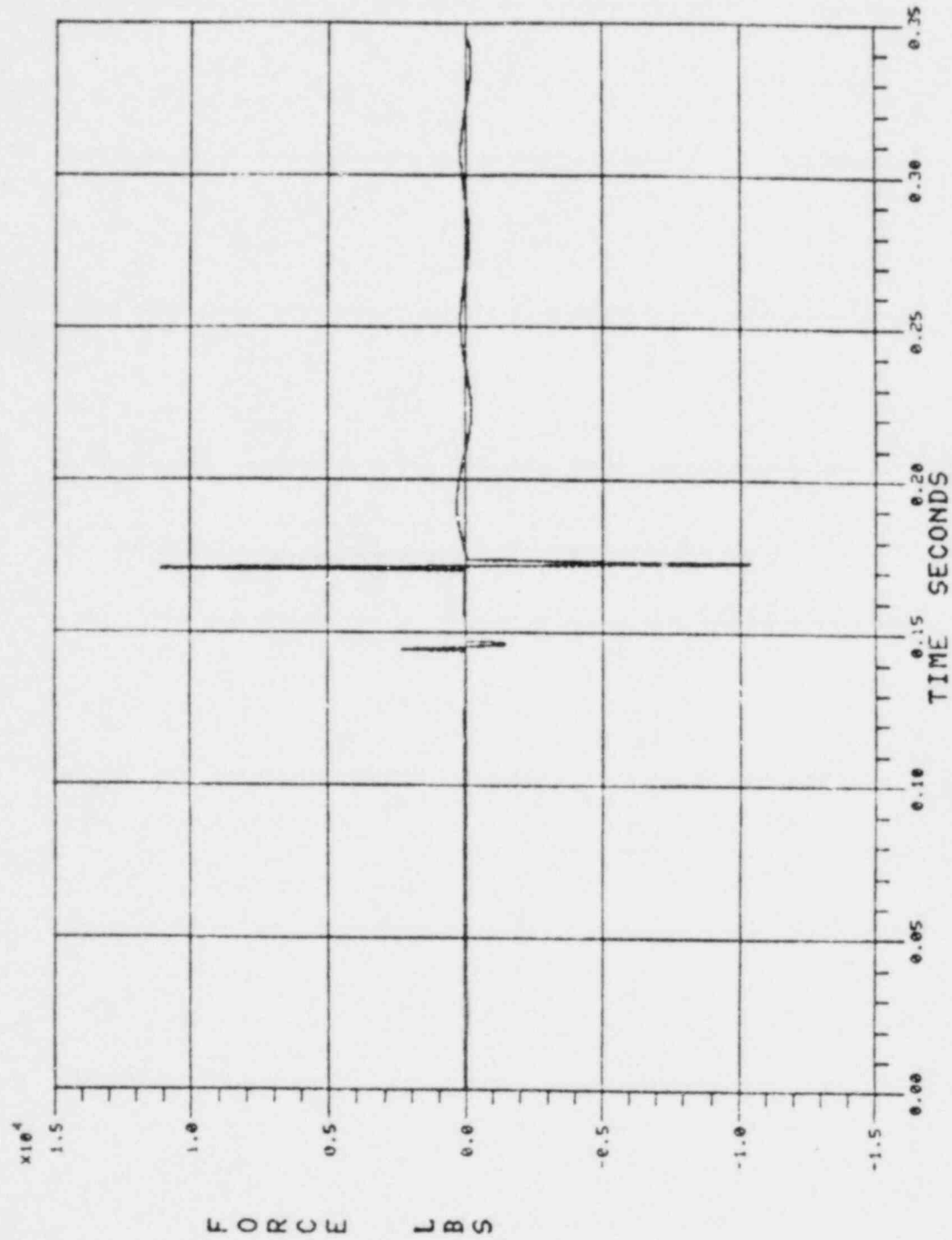
Figure 2.21-3. (Cont'd)



2.21-37

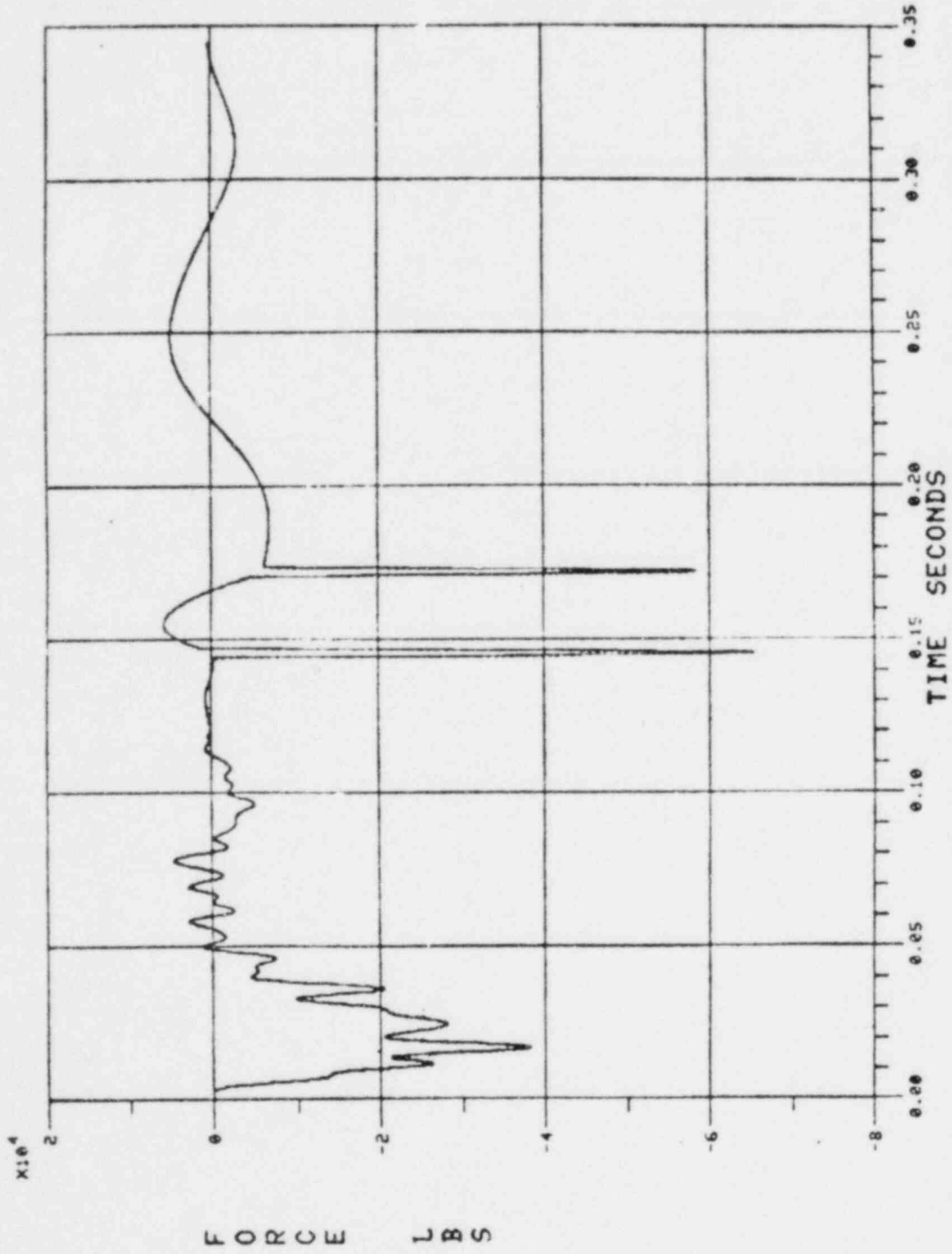
JOINT 62 Z DIR CORE SUPPORT SHEILD

Figure 2.21-3. (Cont'd)



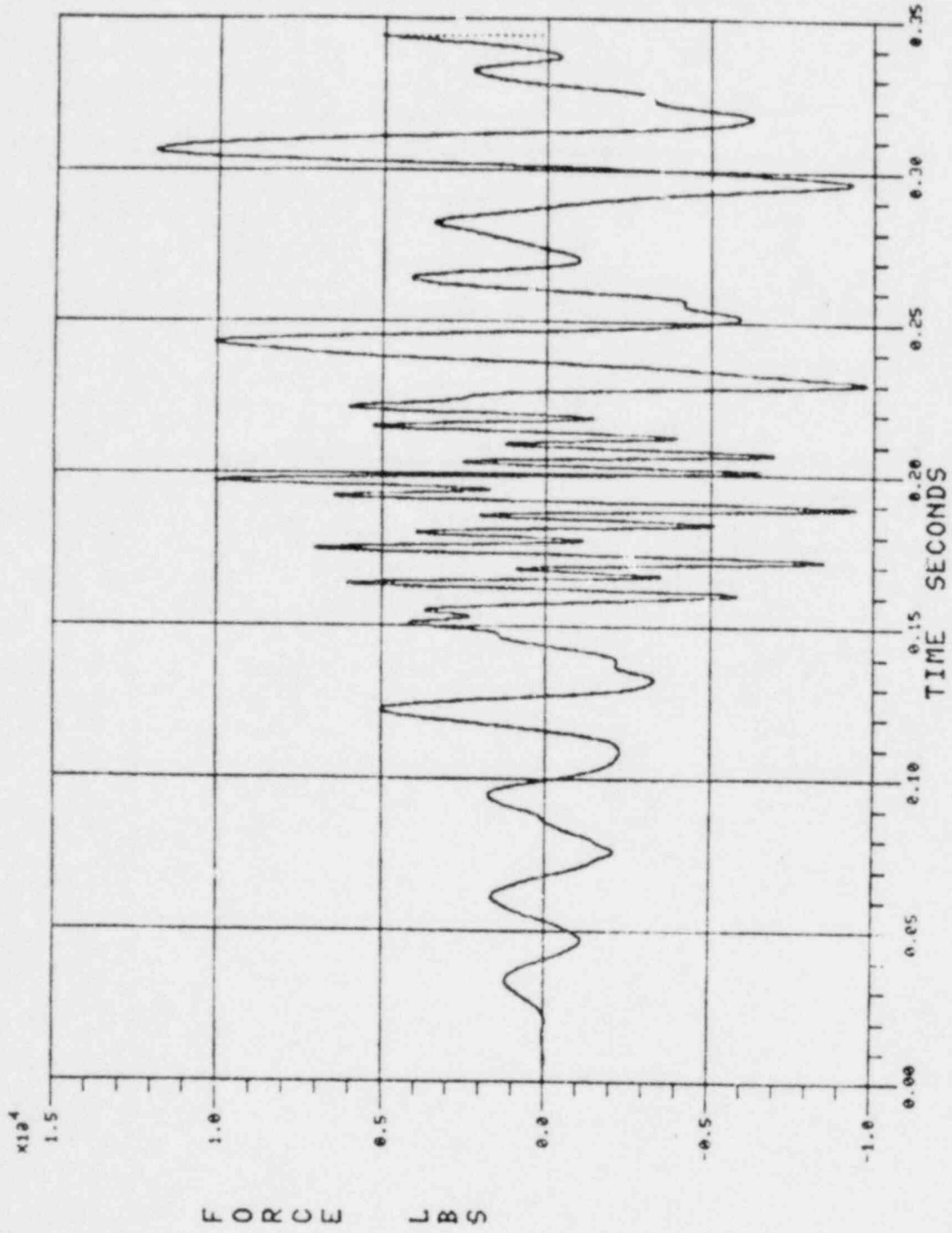
JOINT 53 X DIR CORE SUPPORT SHEILD

Figure 2.21-3. (Cont'd)



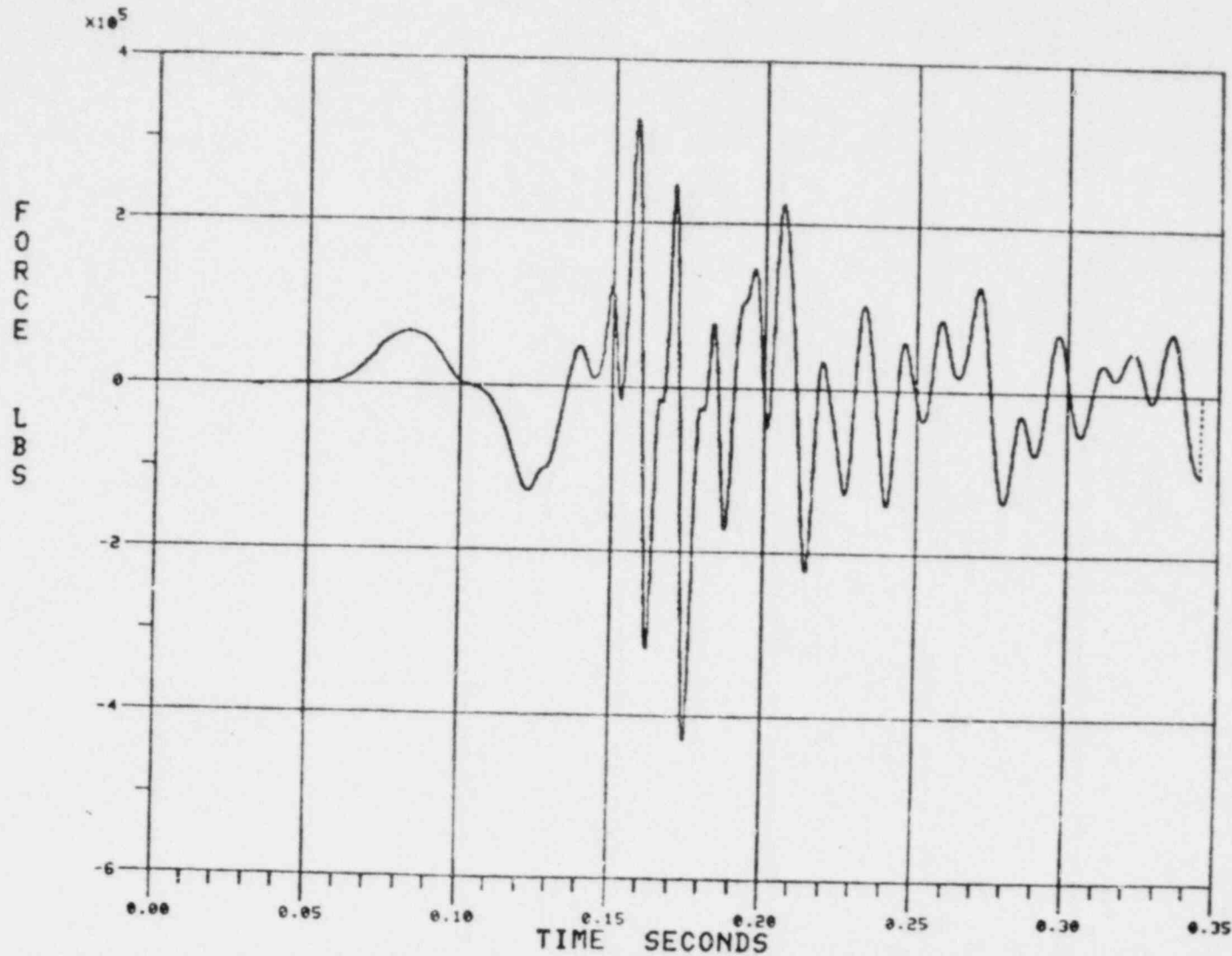
JOINT 53 Z DIR CORE SUPPORT SHEILD

Figure 2.21-3. (Cont'd)



JOINT 7 X DIR THERMALSHIELD

Figure 2.21-3. (Cont'd)

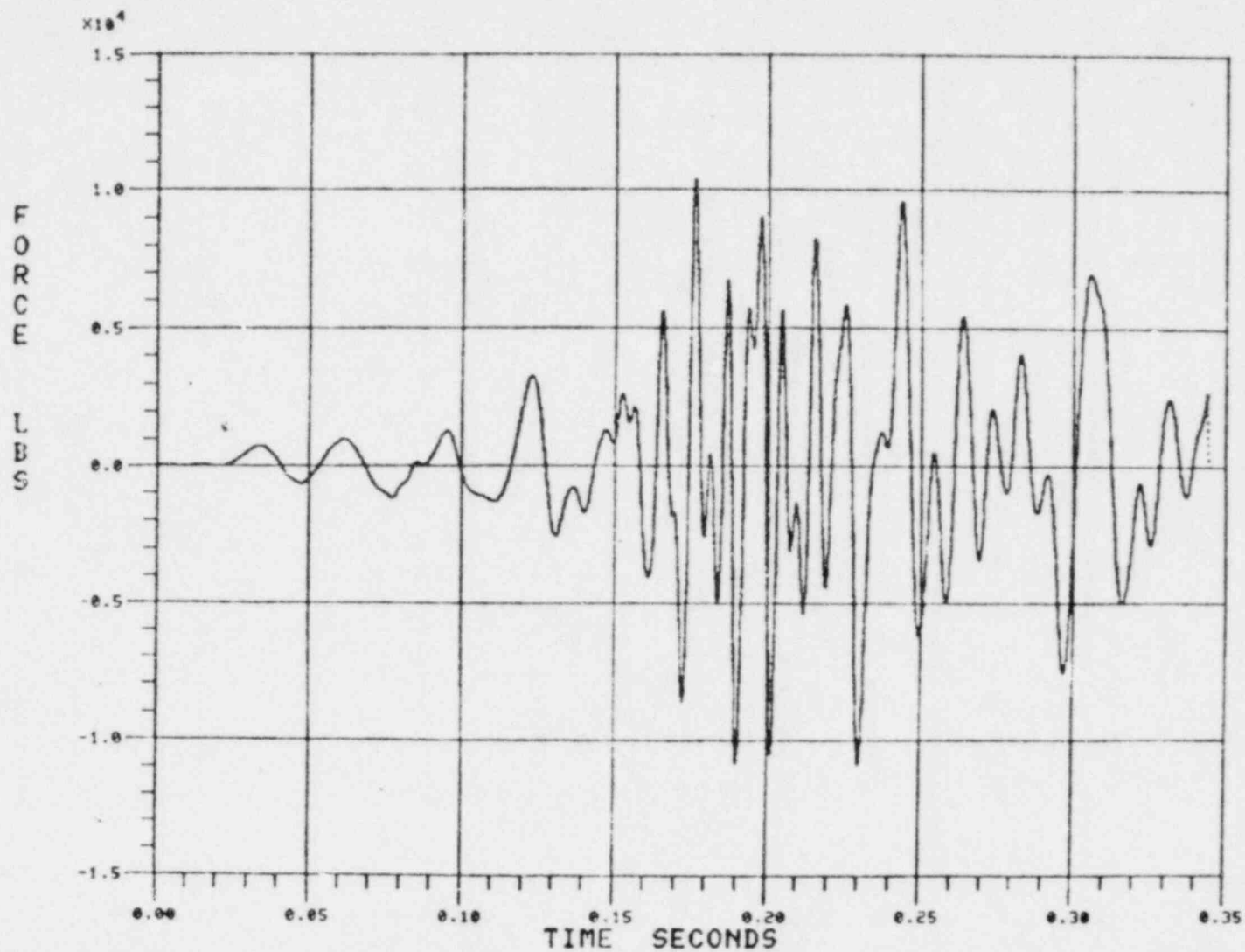


2.21-41

JOINT 7 Z DIR THERMALSHEILD



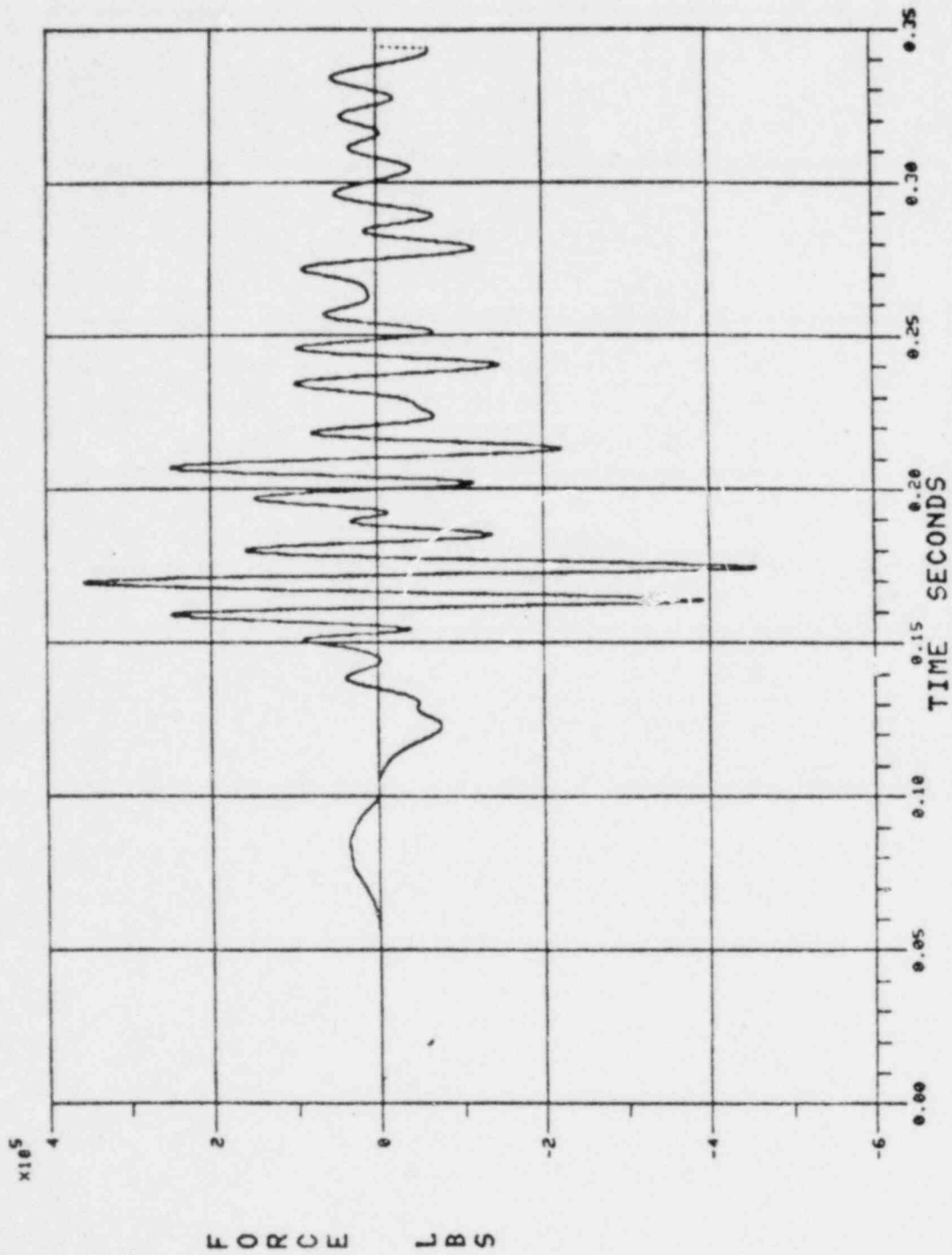
Figure 2.21-3. (Cont'd)



2.21-42

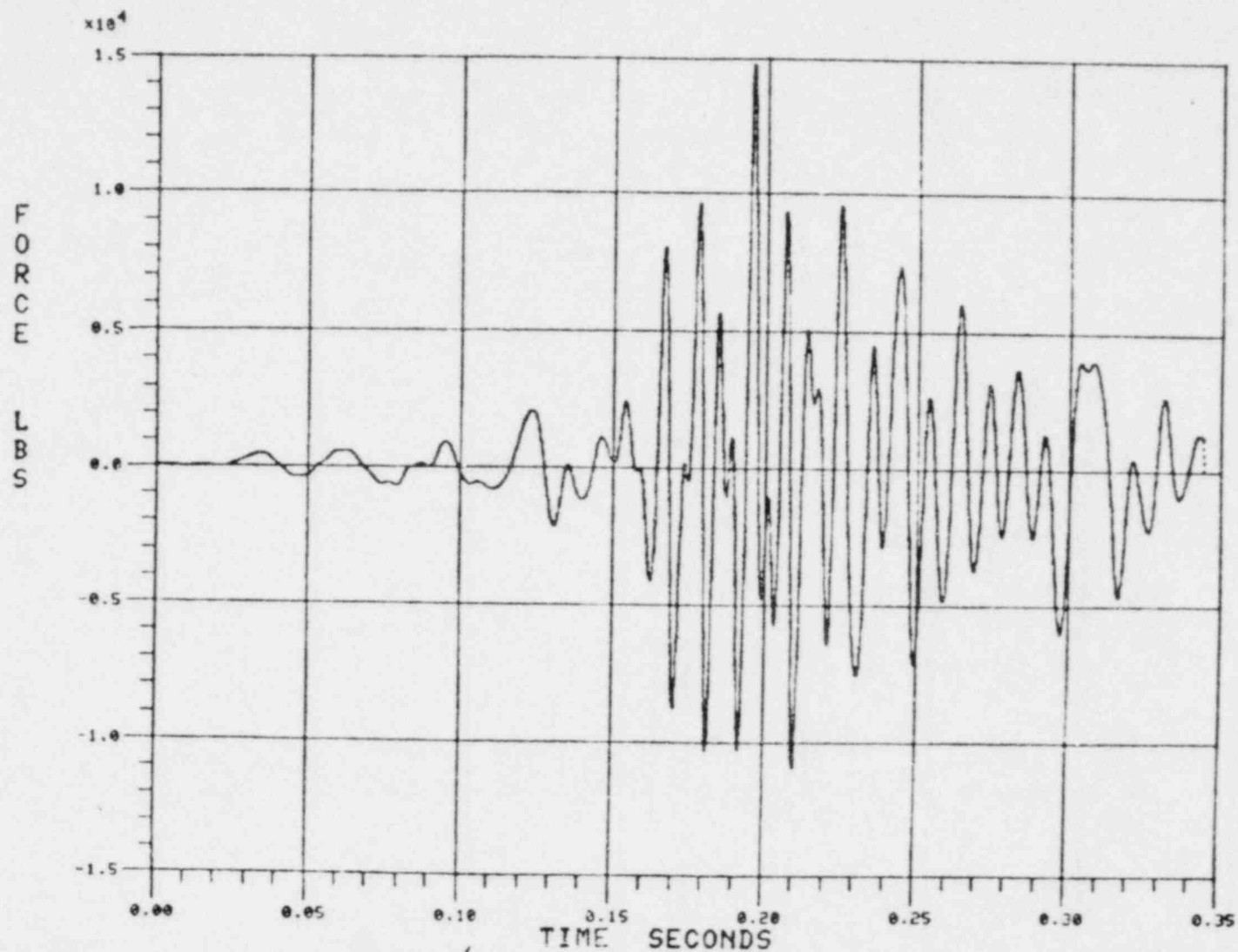
JOINT 1 X DIR THERMALSHEILD

Figure 2.21-3. (Cont'd)



JOINT 1 Z DIR THERMALSHEILD

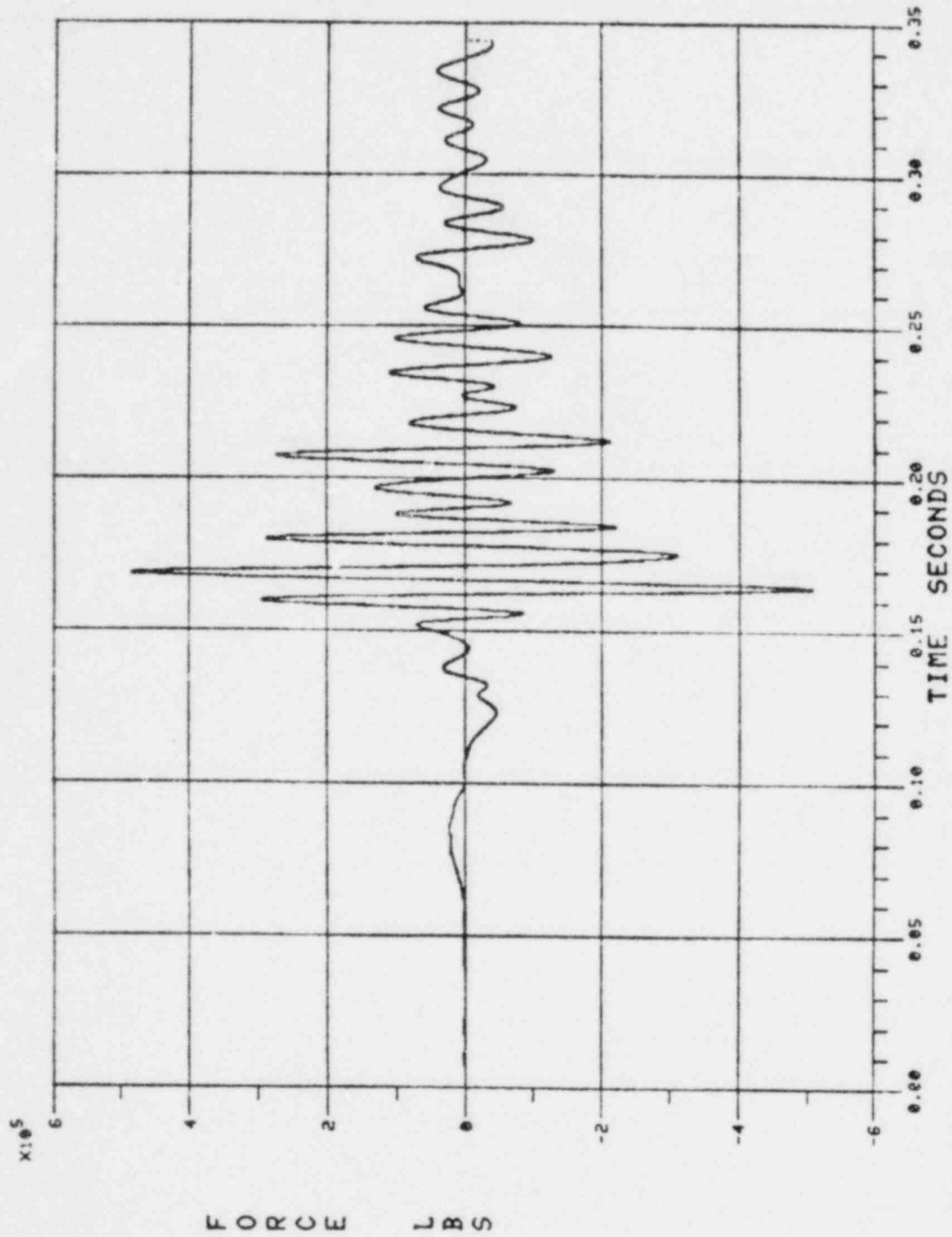
Figure 2.21-3. (Cont'd)



2.21-44

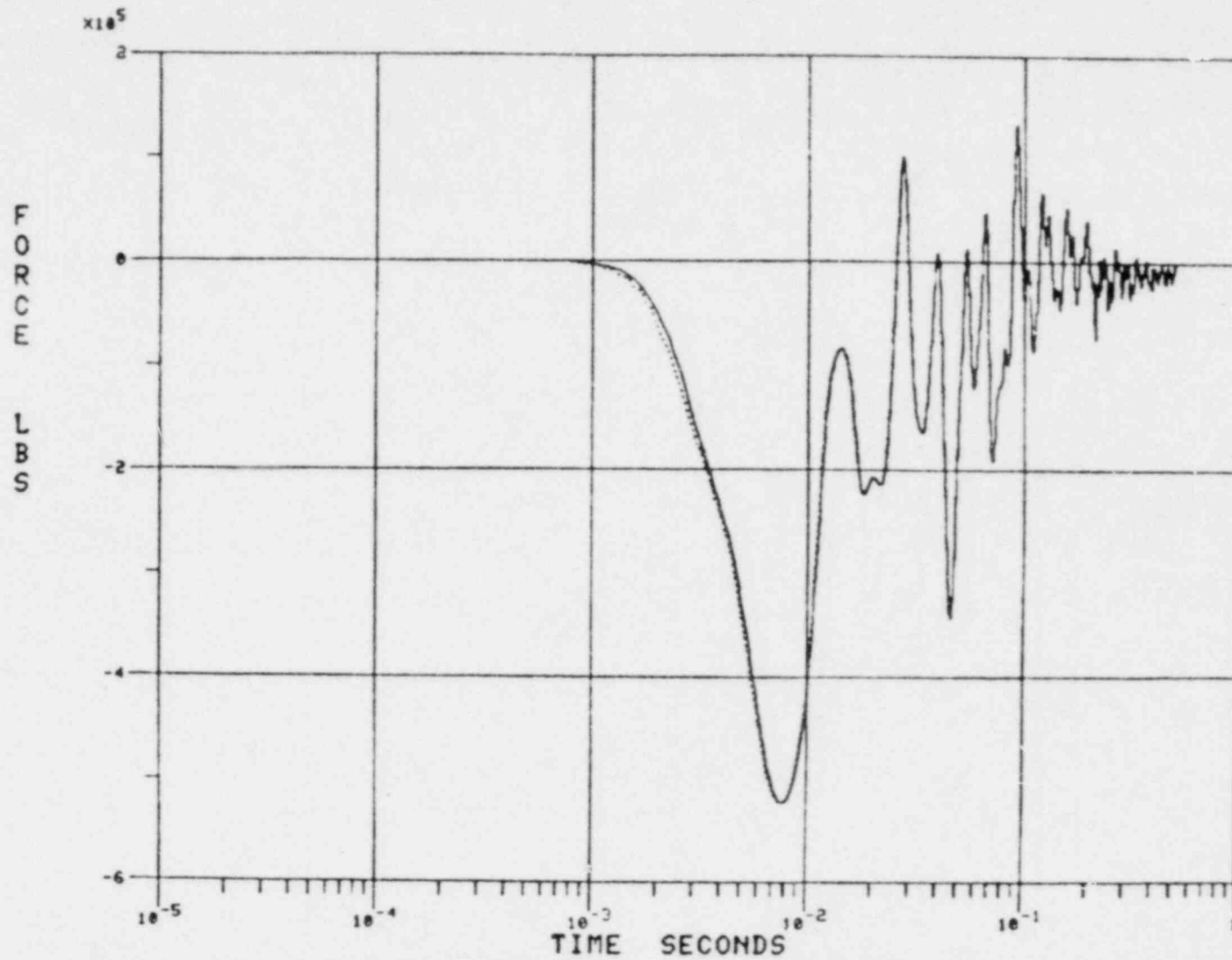
JOINT 6 X DIR THERMALSHEILD

Figure 2.21-3. (Cont'd)



JOINT 6 Z DIR THERMALSHEILD

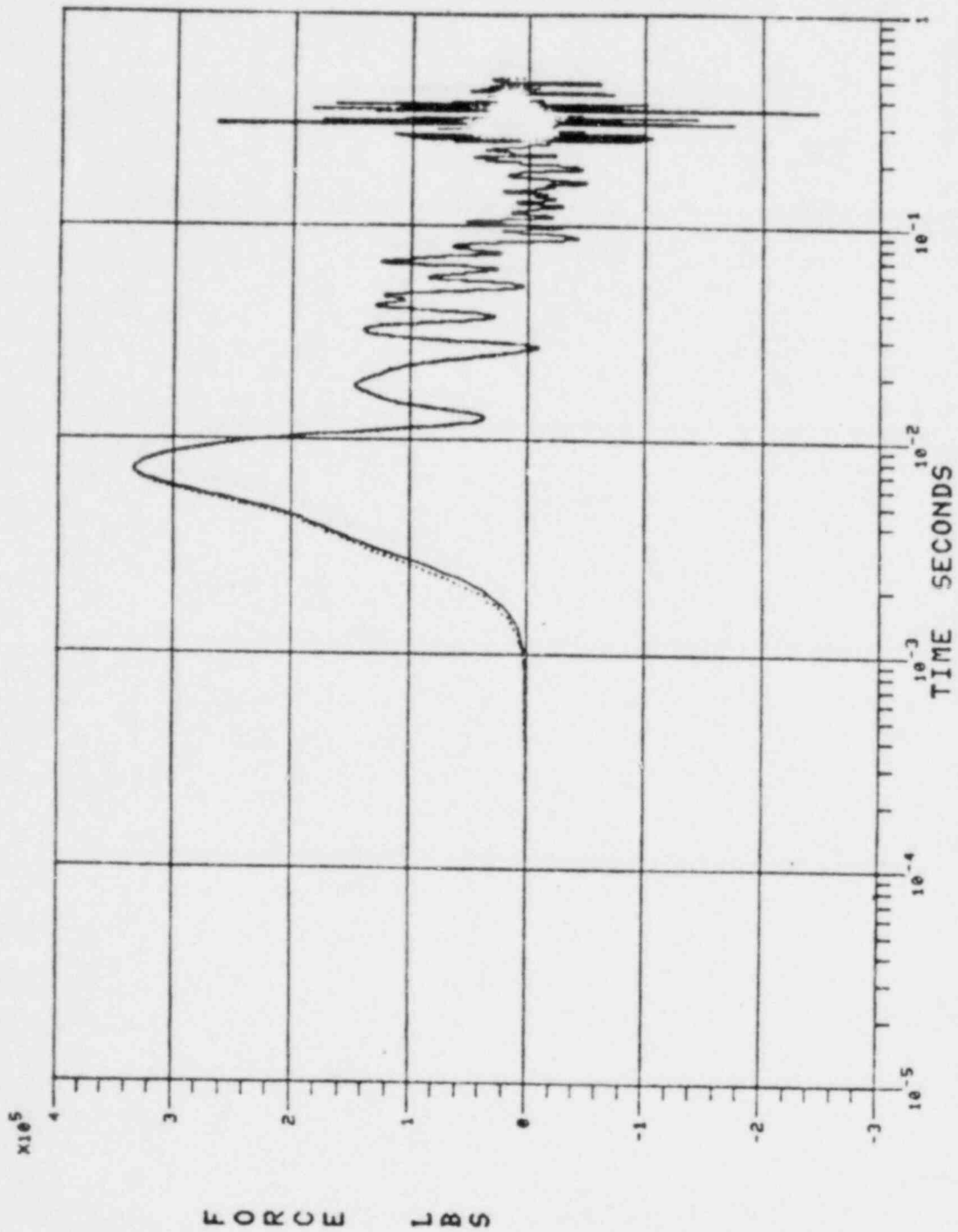
Figure 2.21-4. Force Across Core Support Shield - 0.242A Cold Leg Break at RV, Nozzle-Supported Plant



2.21-46

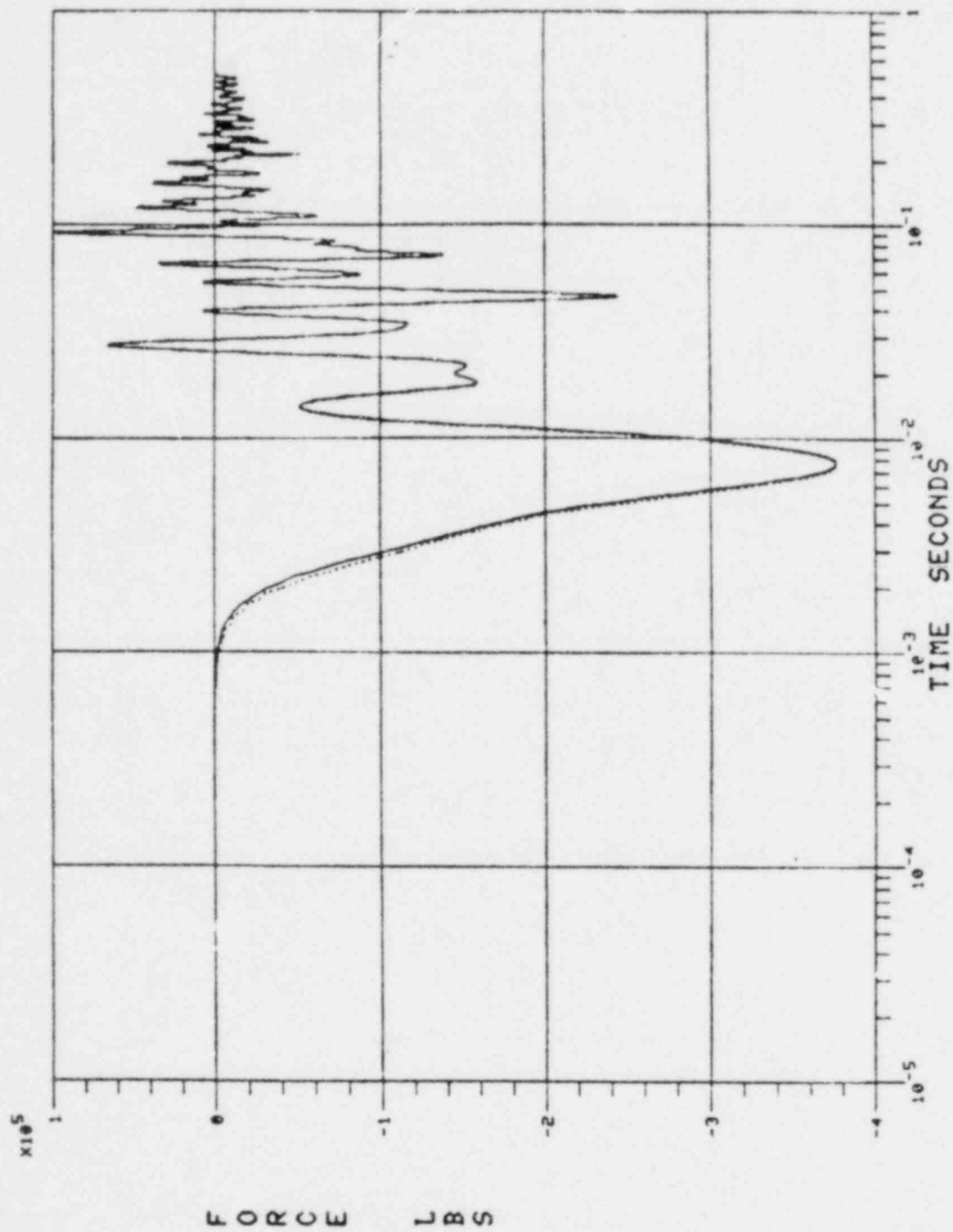
JOINT 31 X DIR CORE SUPPORT SHEILD

Figure 2.21-4. (Cont'd)



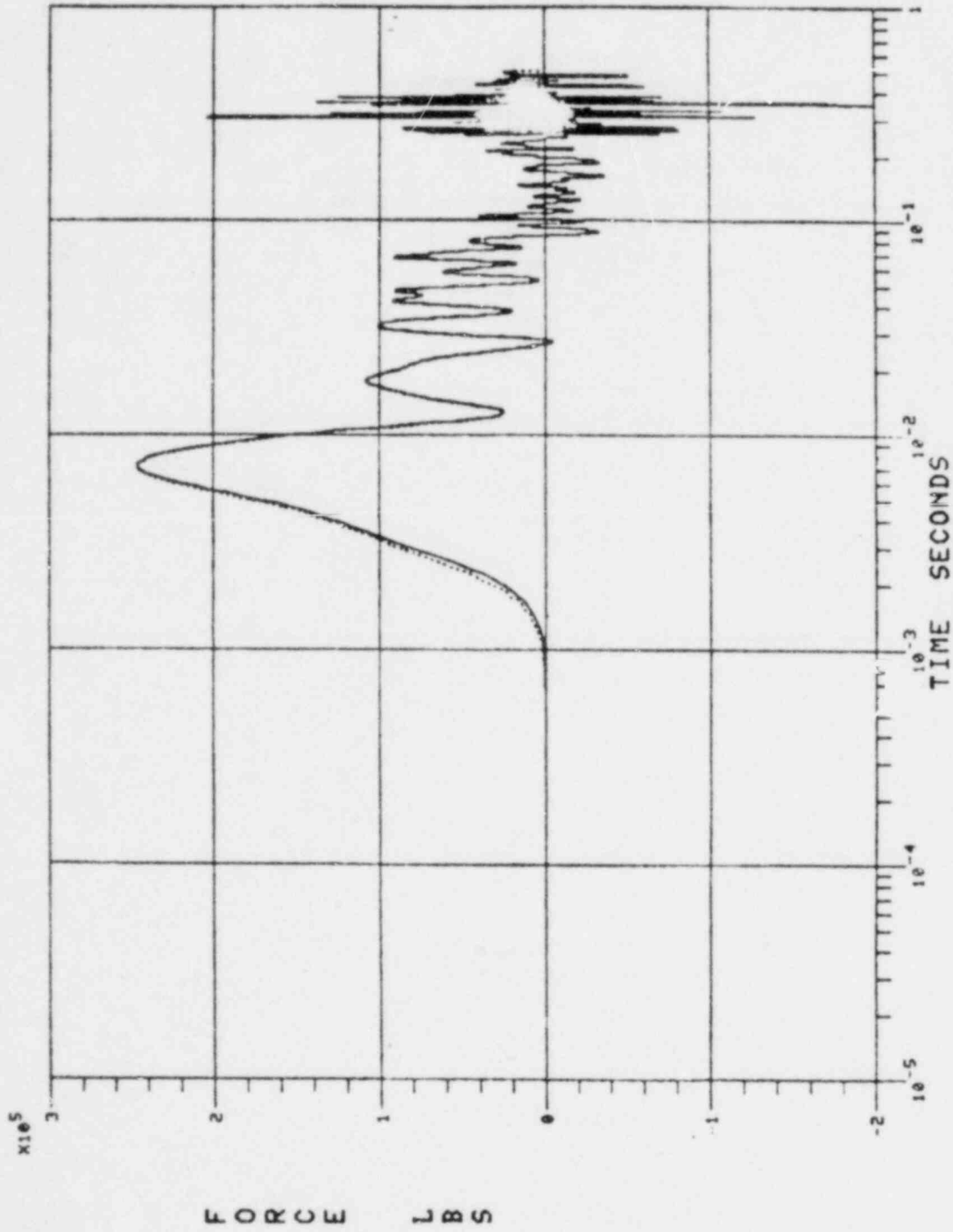
JOINT 31 Z DIR CORE SUPPORT SHEILD

Figure 2.21-4. (Cont'd)



JOINT 44 X DIR CORE SUPPORT SHEILD

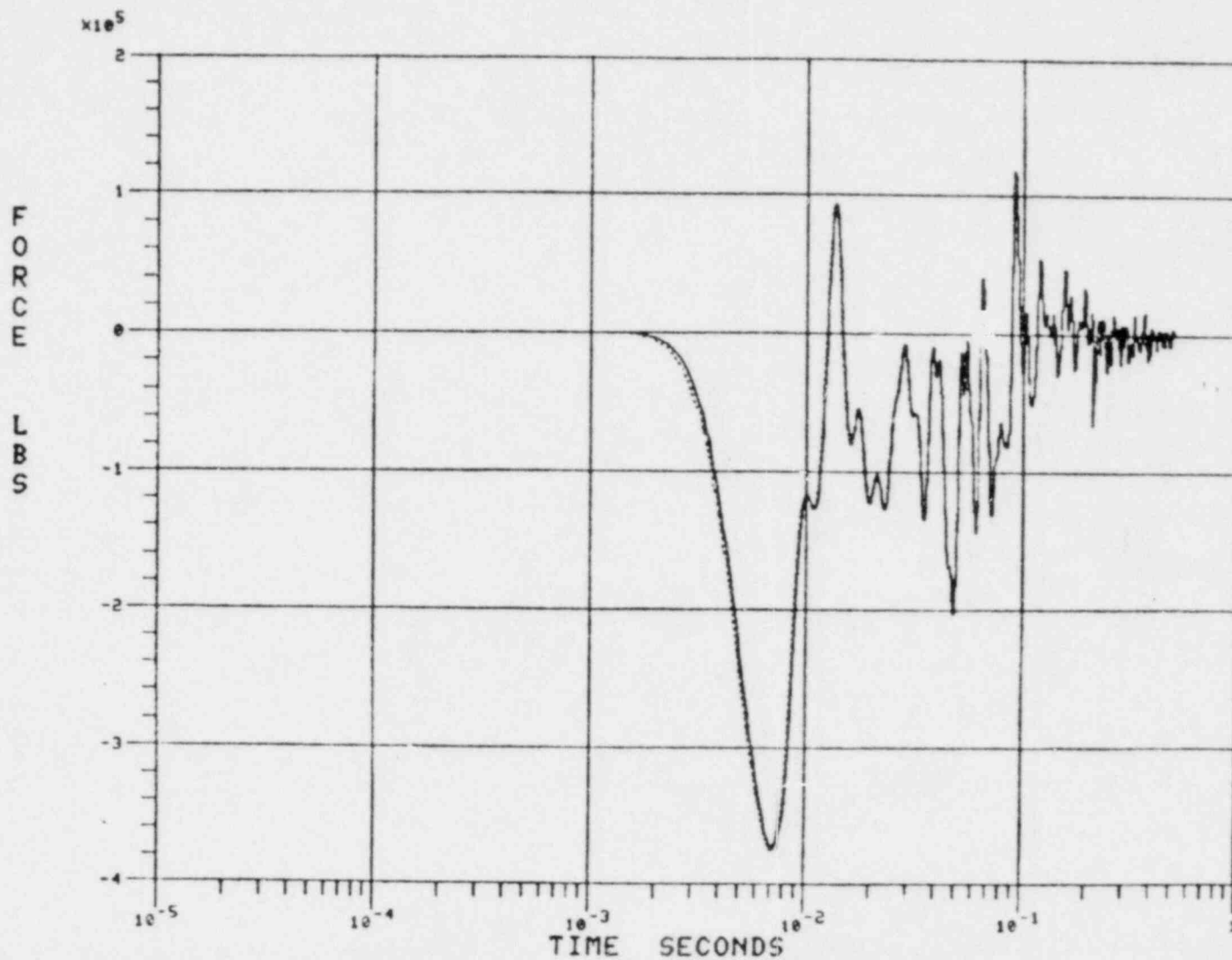
Figure 2.21-4. (Cont'd)



JOINT 44 Z DIR CORE SUPPORT SHEILD



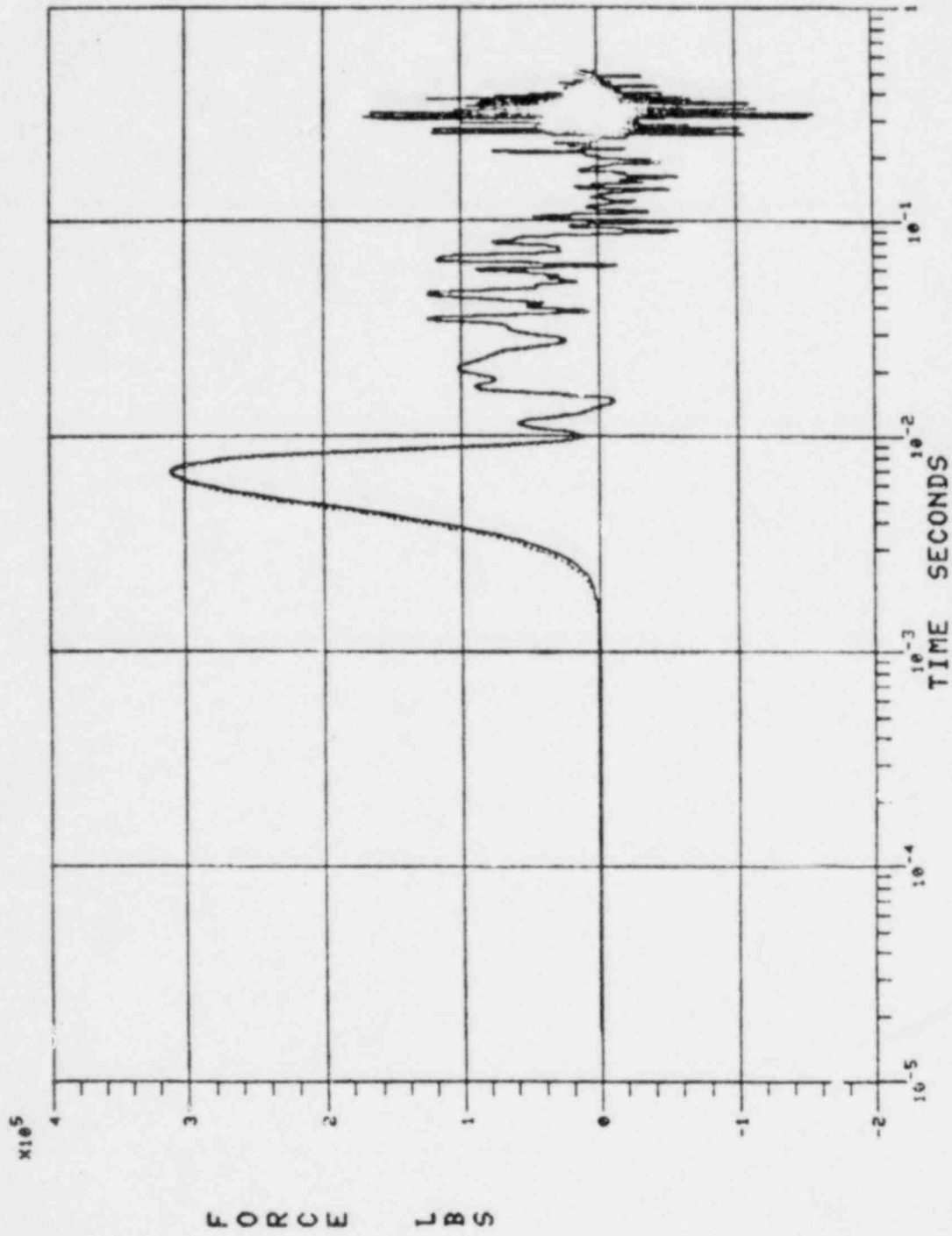
Figure 2.21-4. (Cont'd)



2.21-50

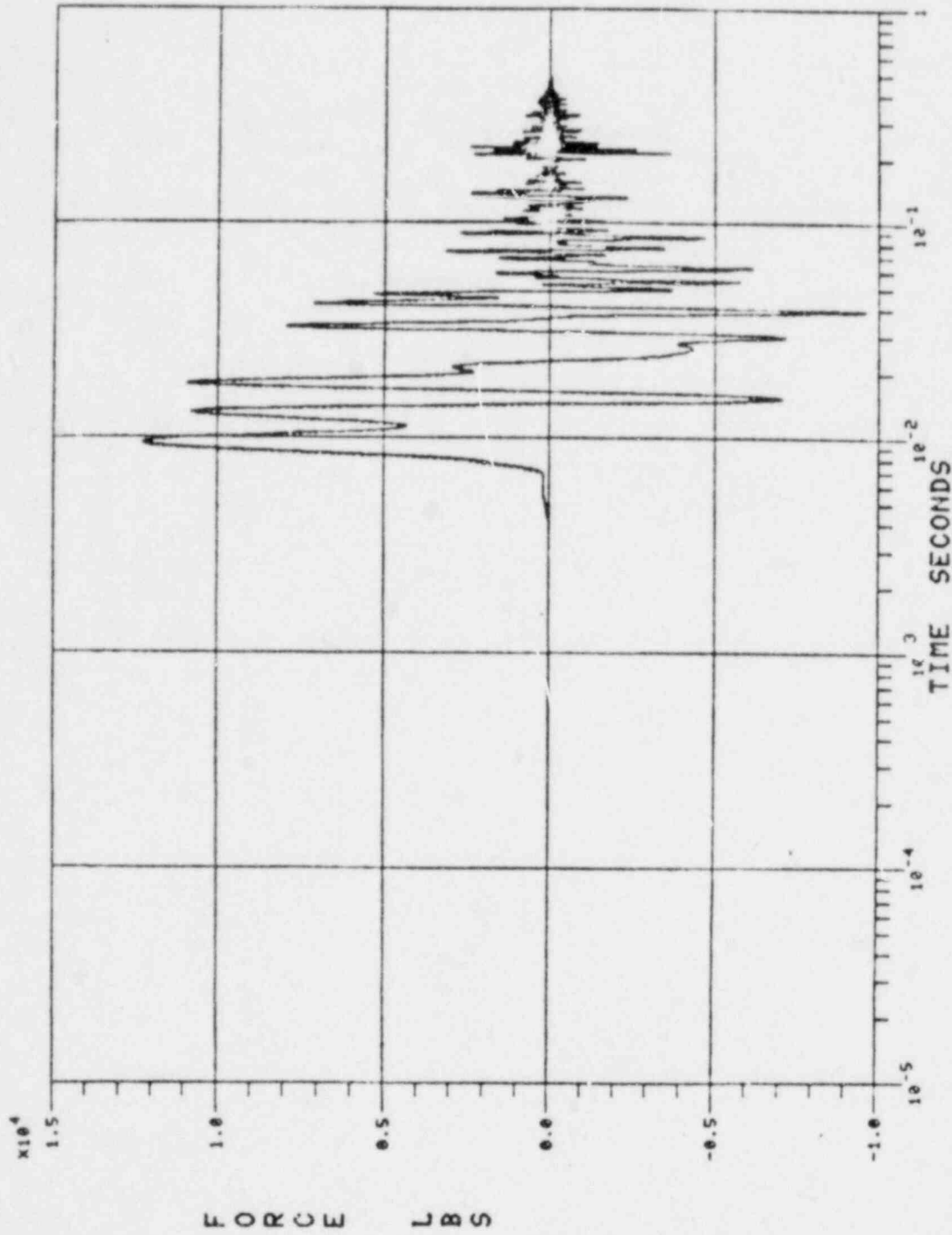
JOINT 62 X DIR CORE SUPPORT SHEILD

Figure 2.21-4. (Cont'd)



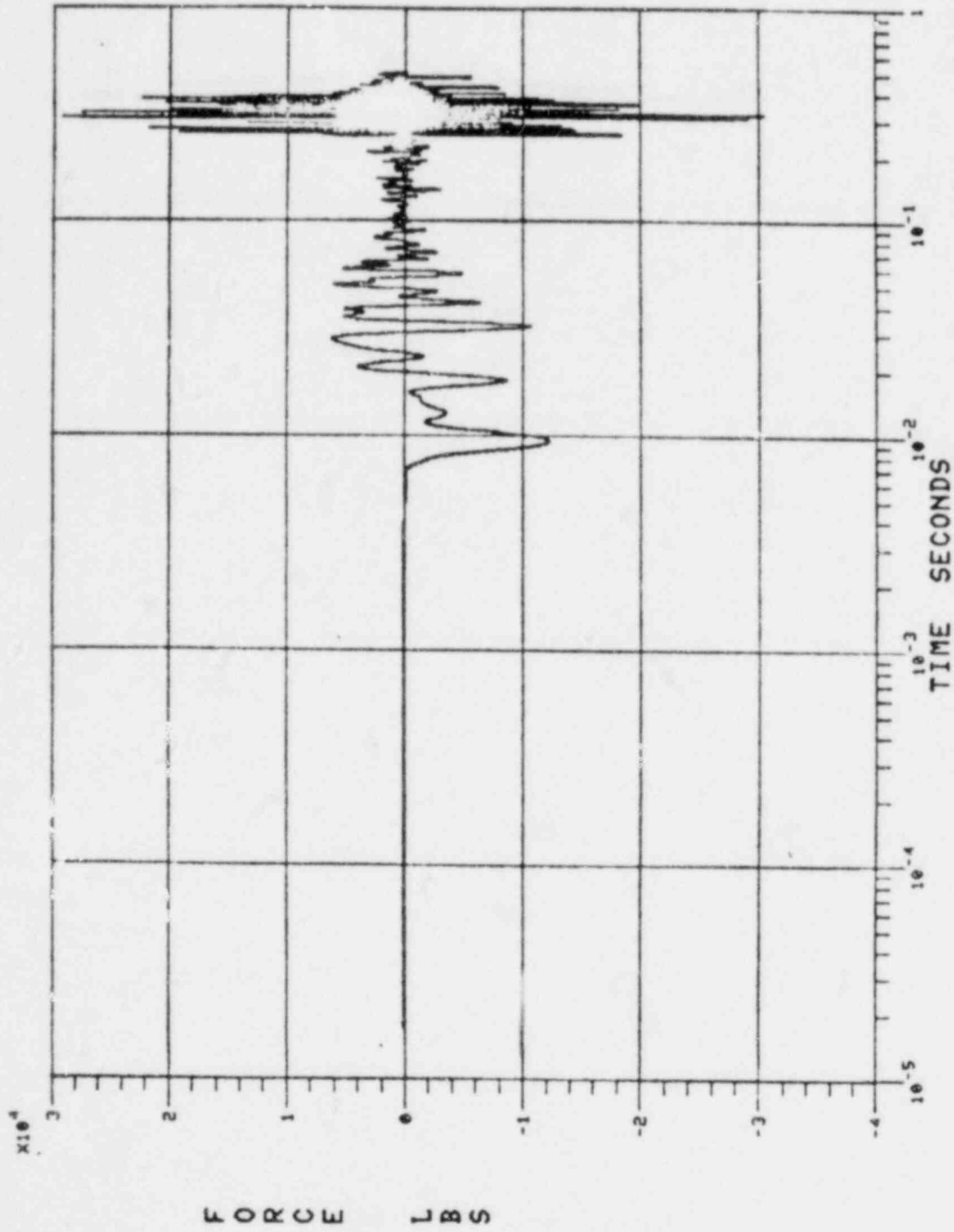
JOINT 62 Z DIR CORE SUPPORT SHIELD

Figure 2.21-4. (Cont'd)



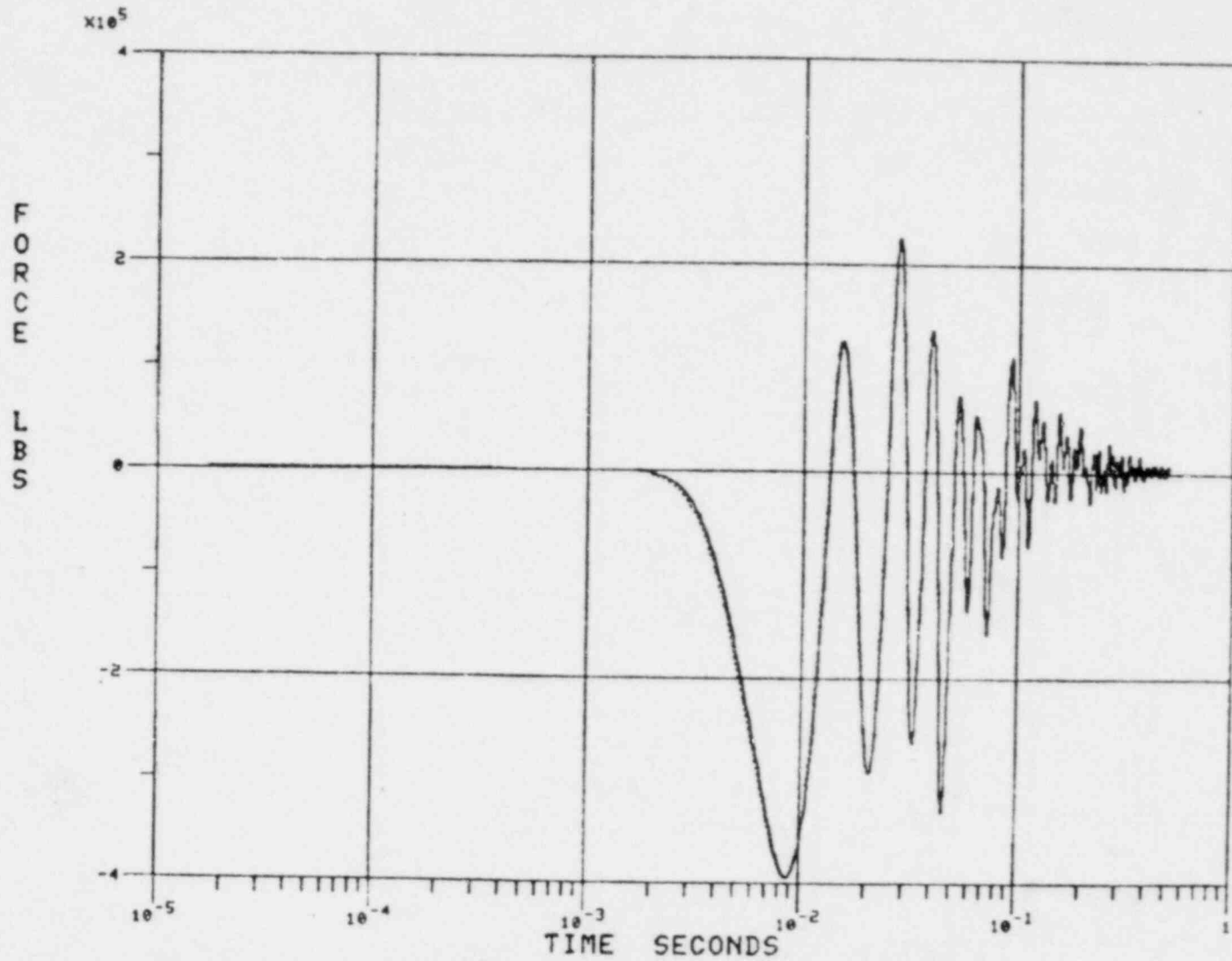
JOINT 53 X DIR CORE SUPPORT SHEILD

Figure 2.21-4. (Cont'd)



JOINT 53 Z DIR CORE SUPPORT SHEILD

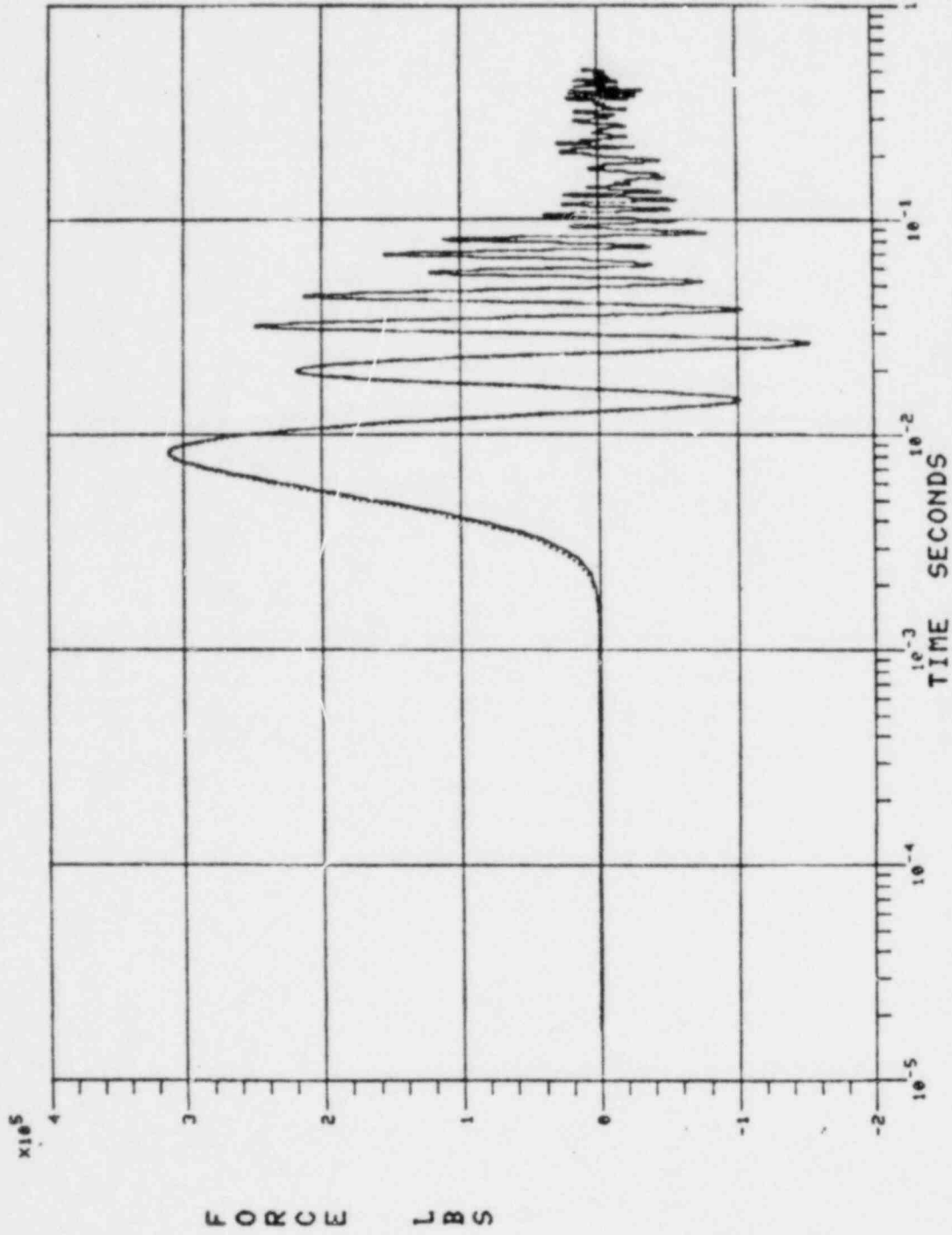
Figure 2.21-4. (Cont'd)



2.21-54

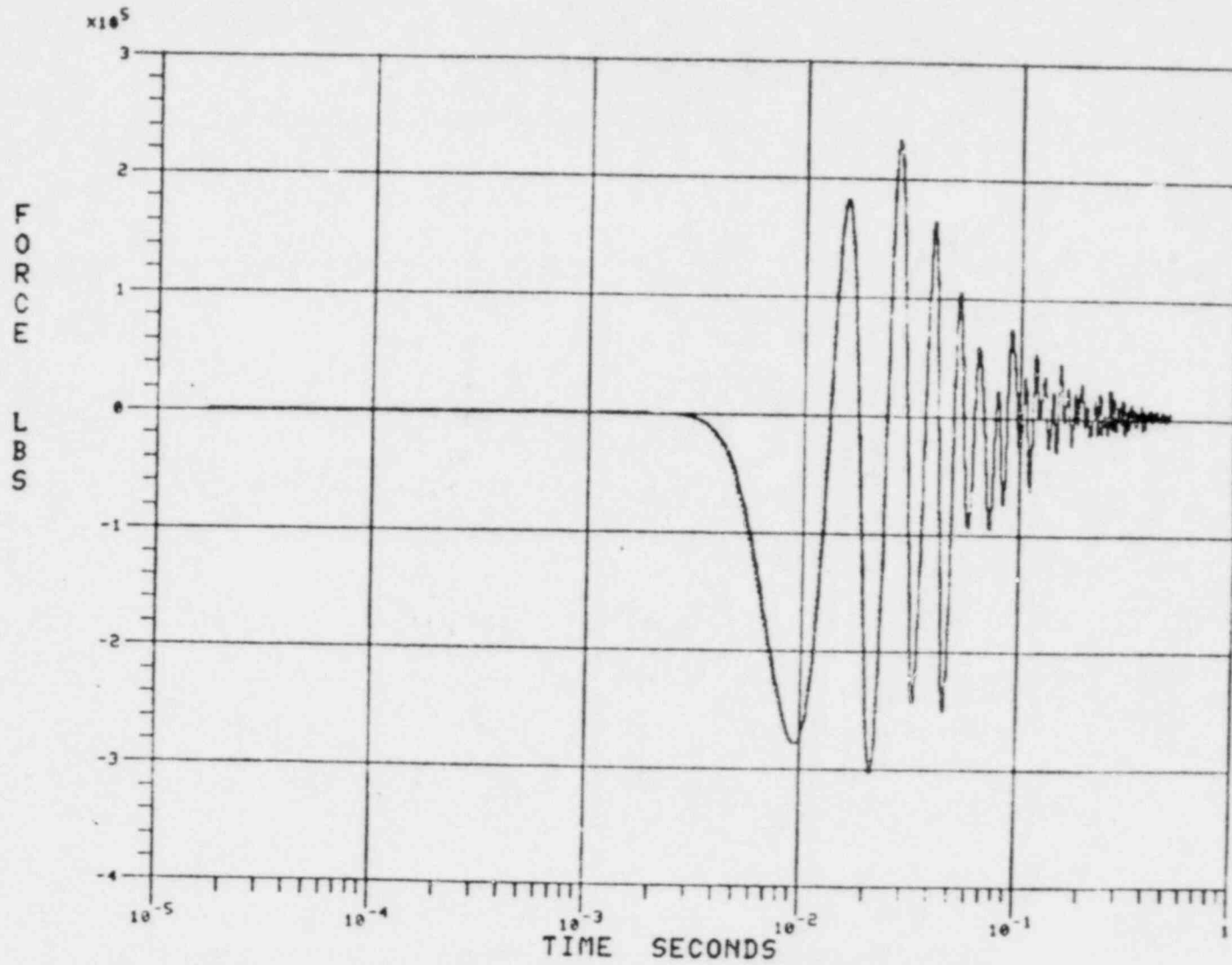
JOINT 7 X DIR THERMALSHIELD

Figure 2.21-4. (Cont'd)



JOINT 7 Z DIR THERMALSHIELD

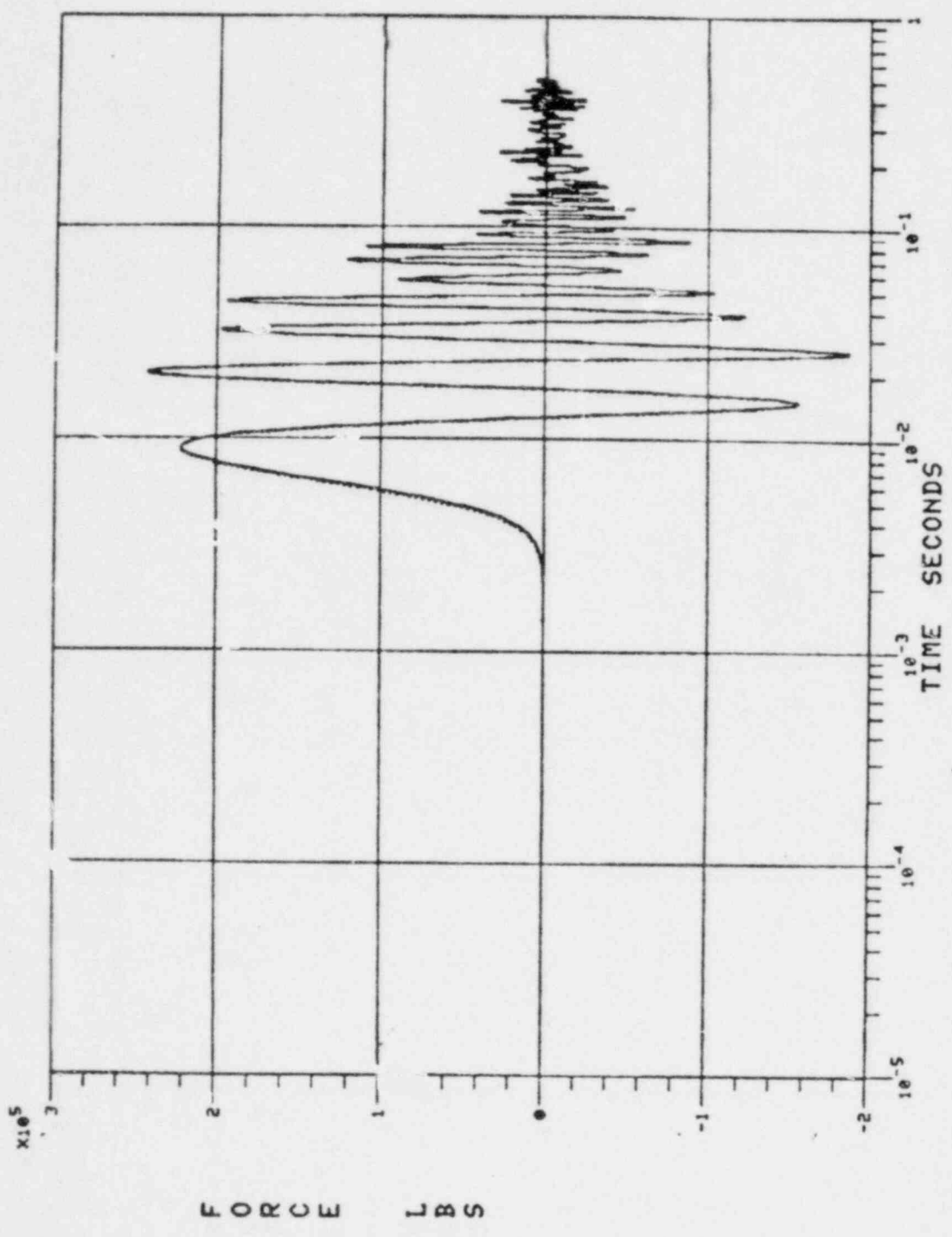
Figure 2.21-4. (Cont'd)



JOINT 1 X DIR THERMALSHIELD

2.21-56

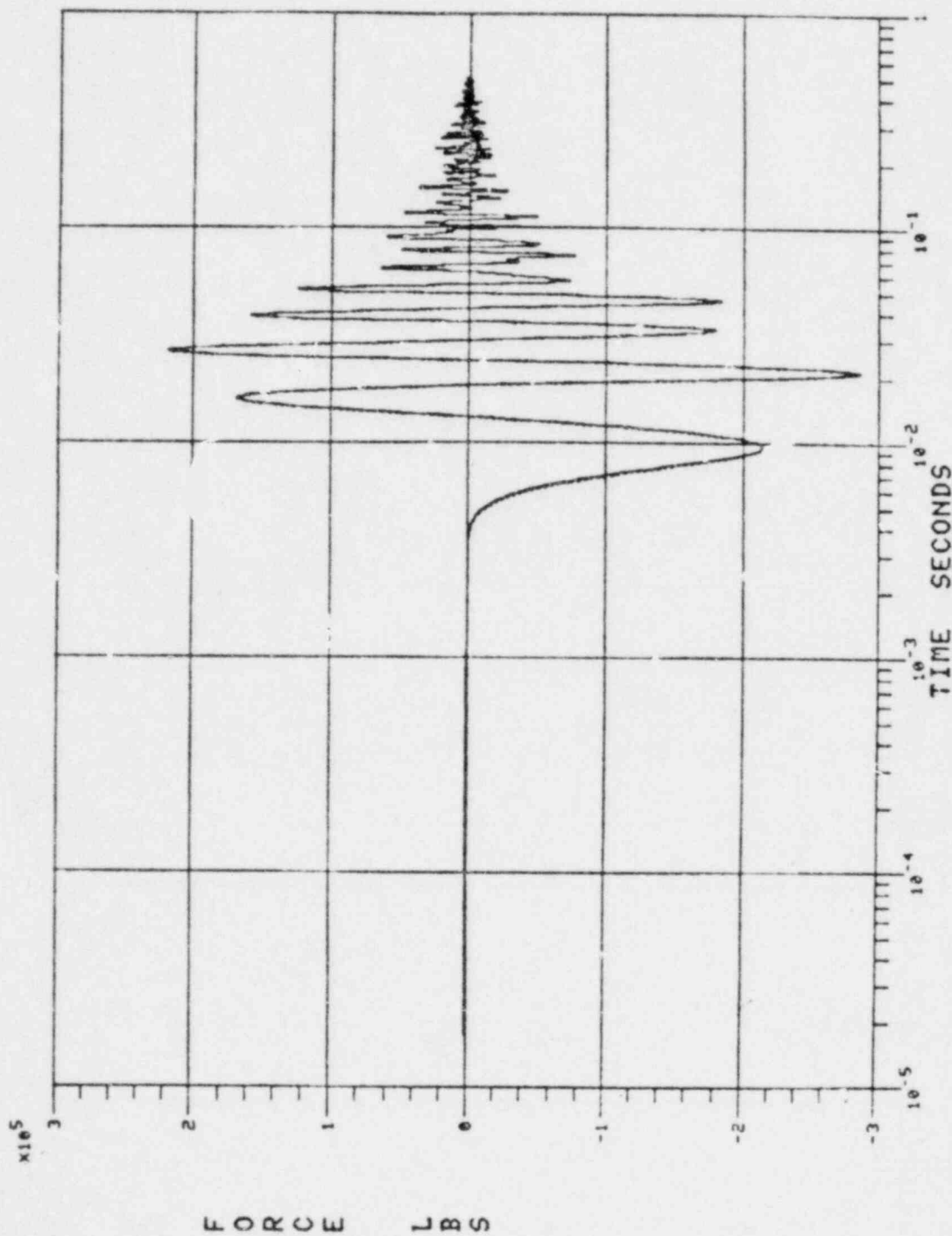
Figure 2.21-4. (Cont'd)



JOINT 1 Z DIR THERMALSHIELD

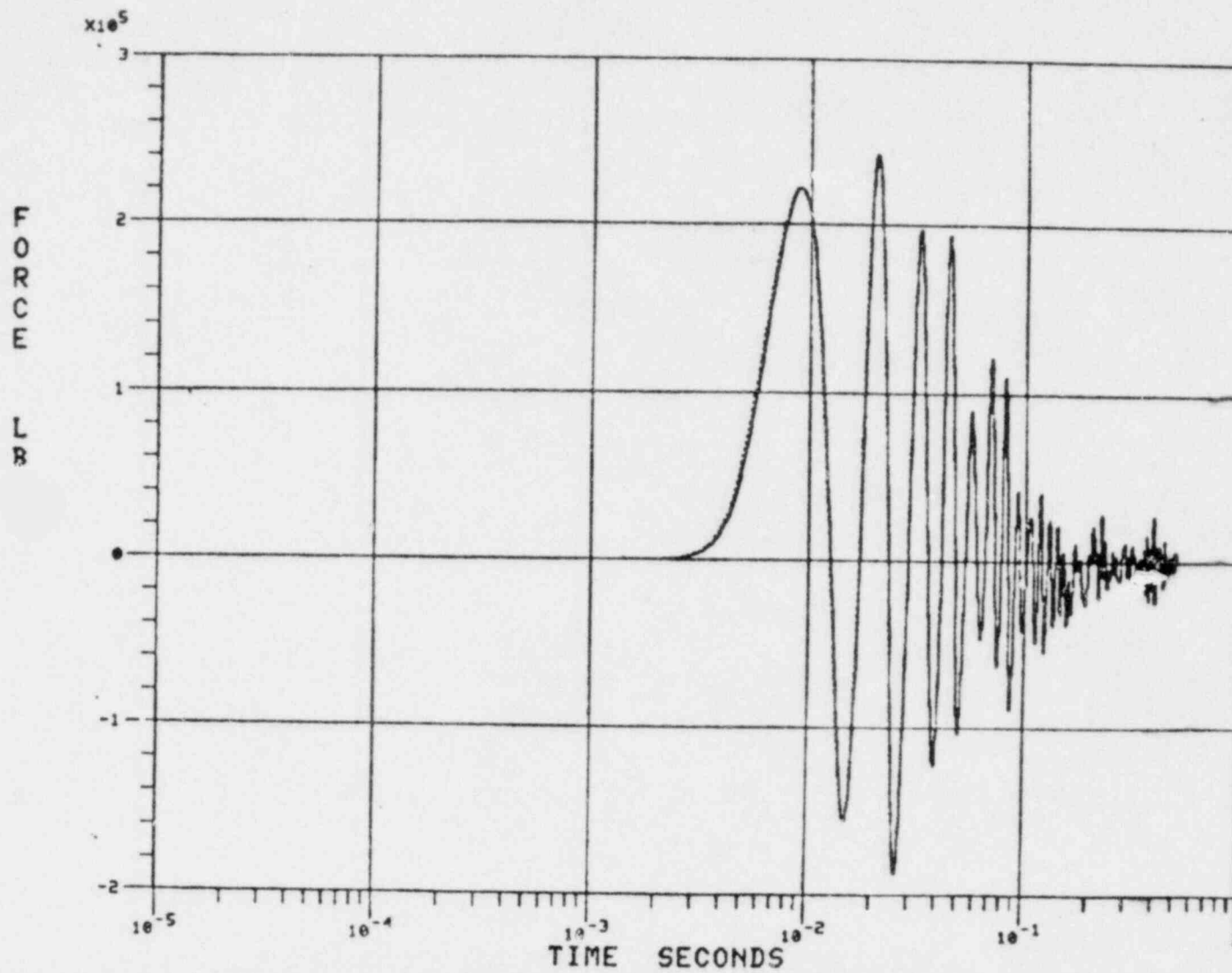


Figure 2.. 1. (Cont'd)



JOINT 6 X DIR THERMALSHEILD

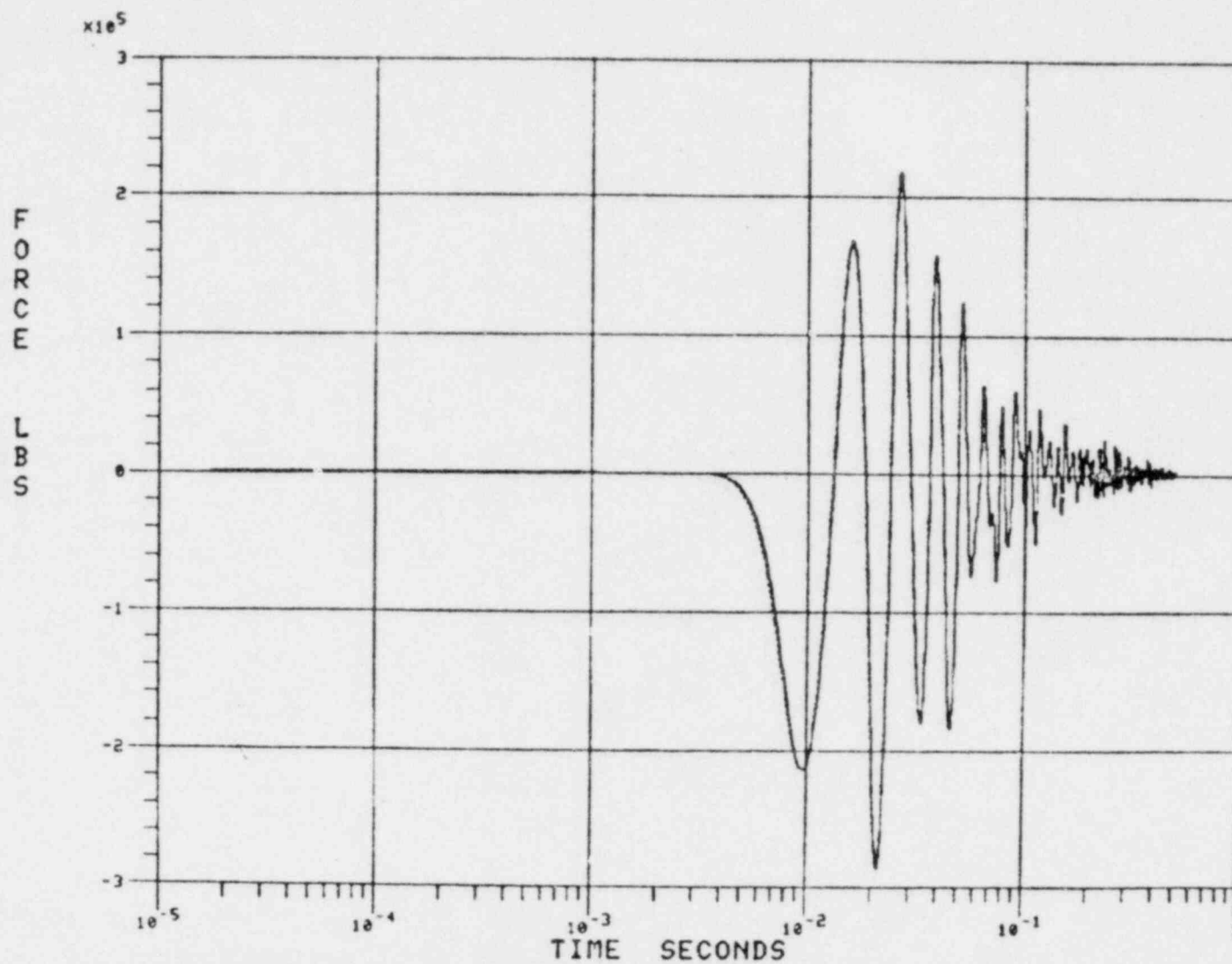
Figure 2.21-4. (Cont'd)



JOINT 1 Z DIR THERMALSHIELD

2.21-57

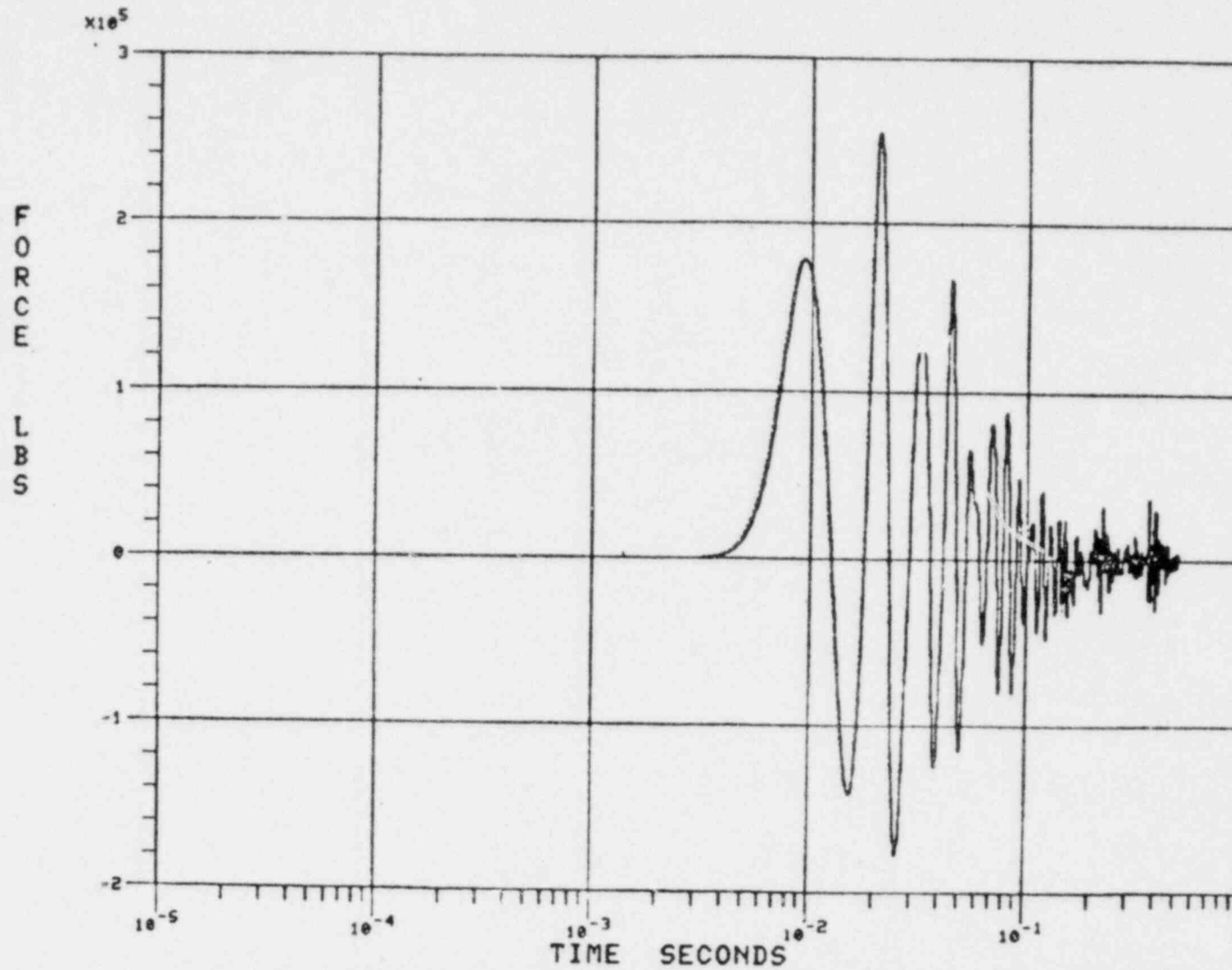
Figure 2.21-4. (Cont'd)



JOINT 6 X DIR THERMALSHEILD

2.21-58

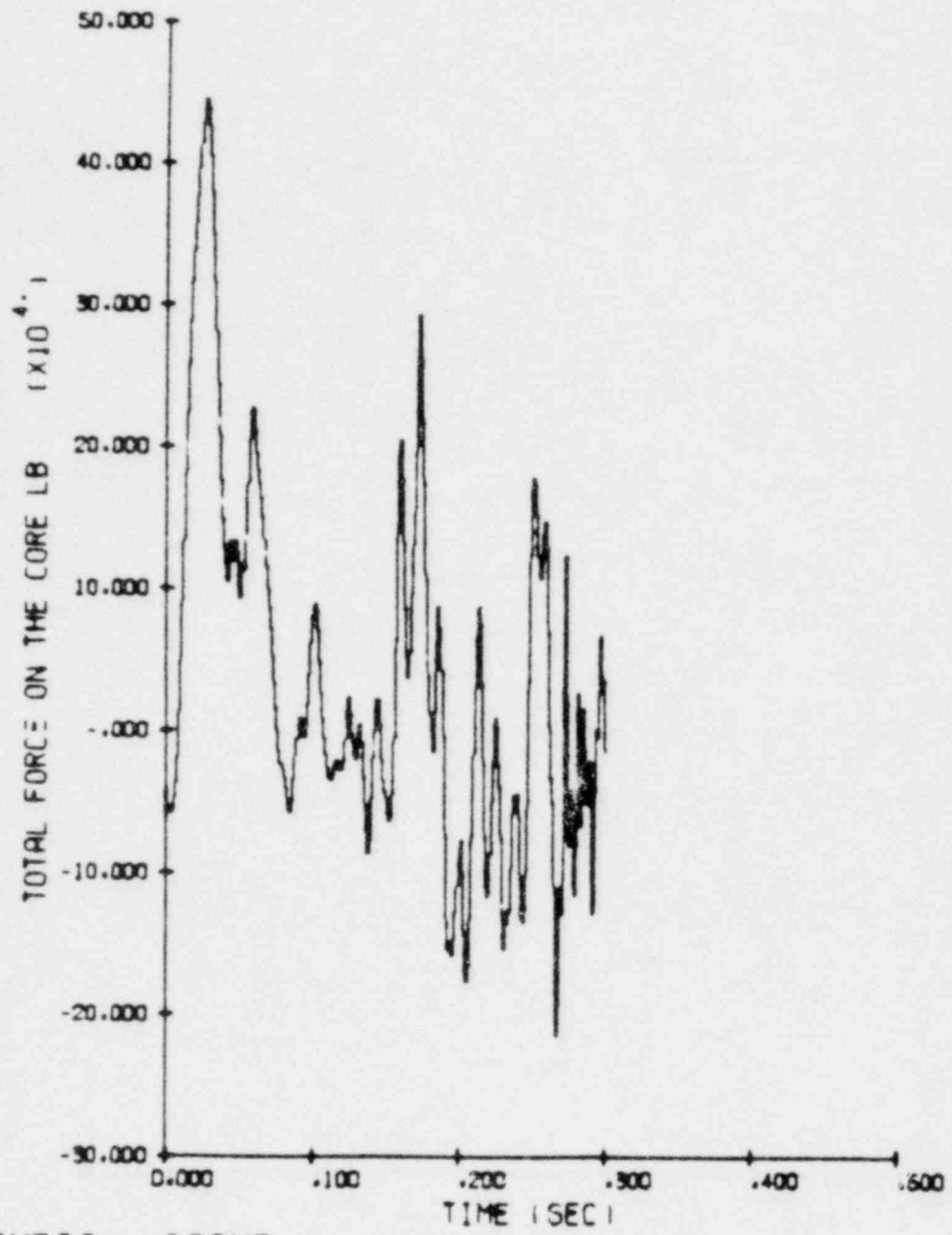
Figure 2.21-4. (Cont'd)



2.21-59

JOINT 6 Z DIR THERMALSHIELD

Figure 2.21-5. Total Force on Core — 2A Hot Leg Break at RV, Skirt-Supported Plant



G2AHB2S G2AHB

Figure 2.21-6. Total Force on Core - 2A Cold Leg Break at RV, Skirt-Supported Plant

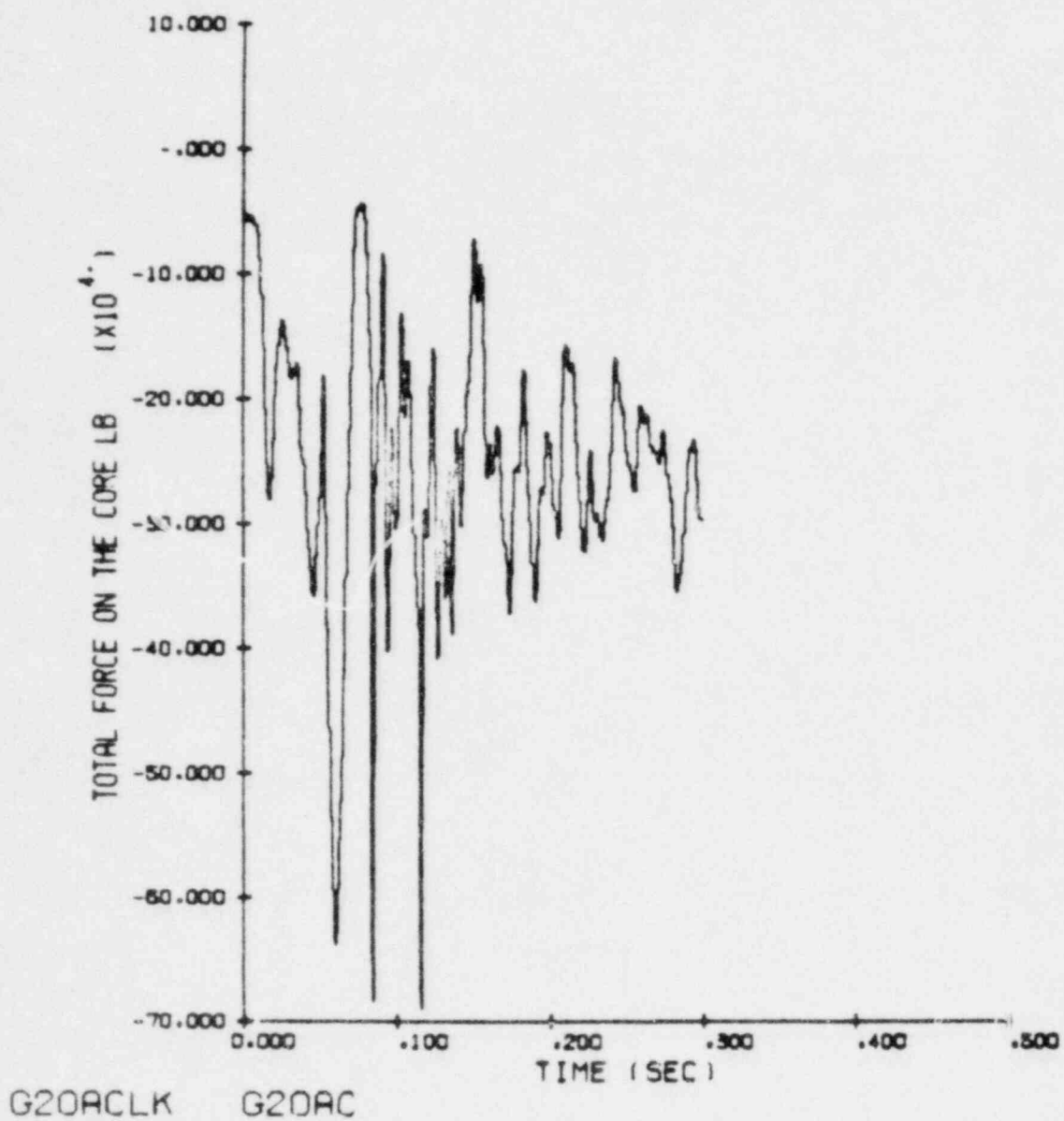
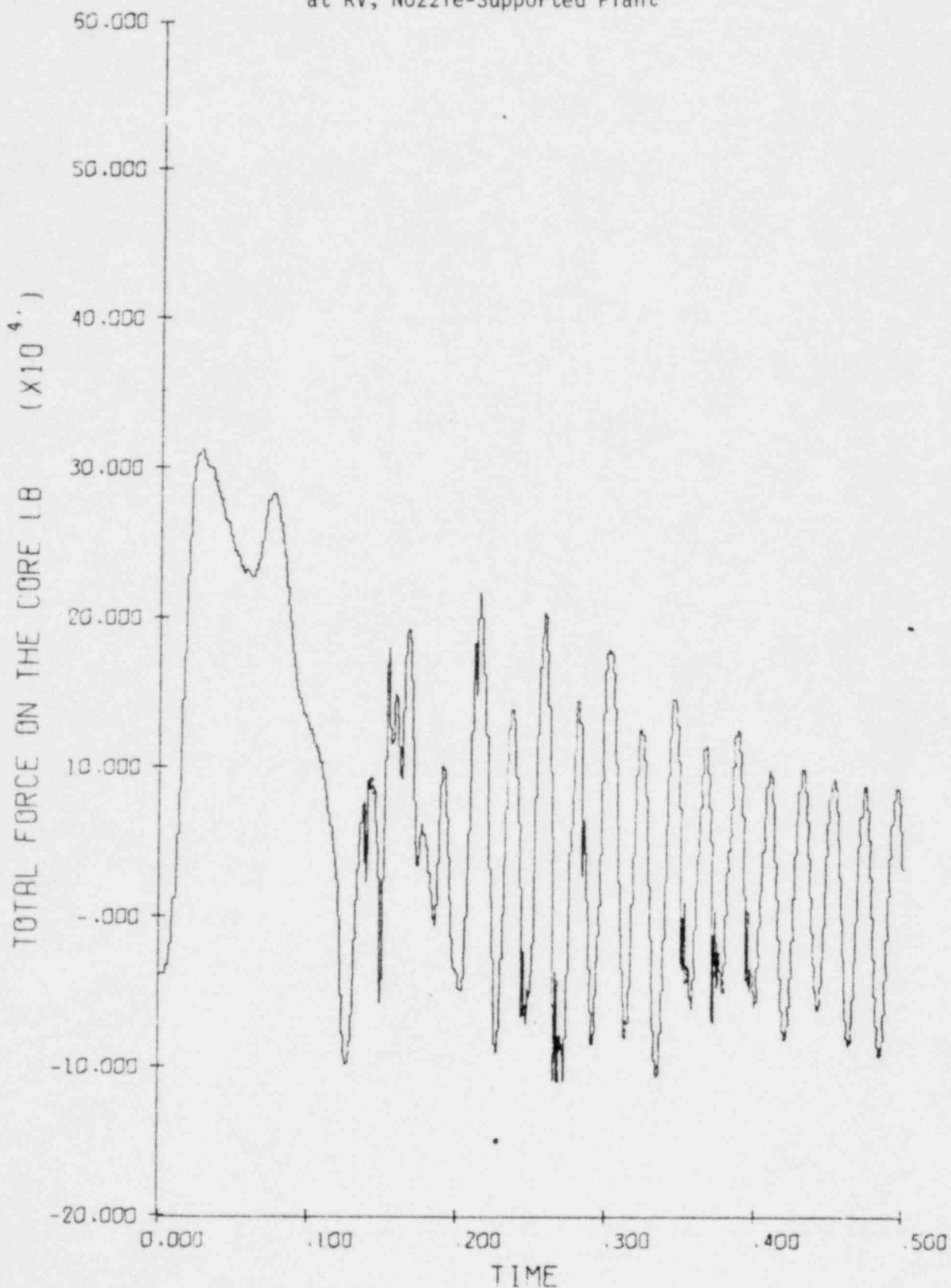


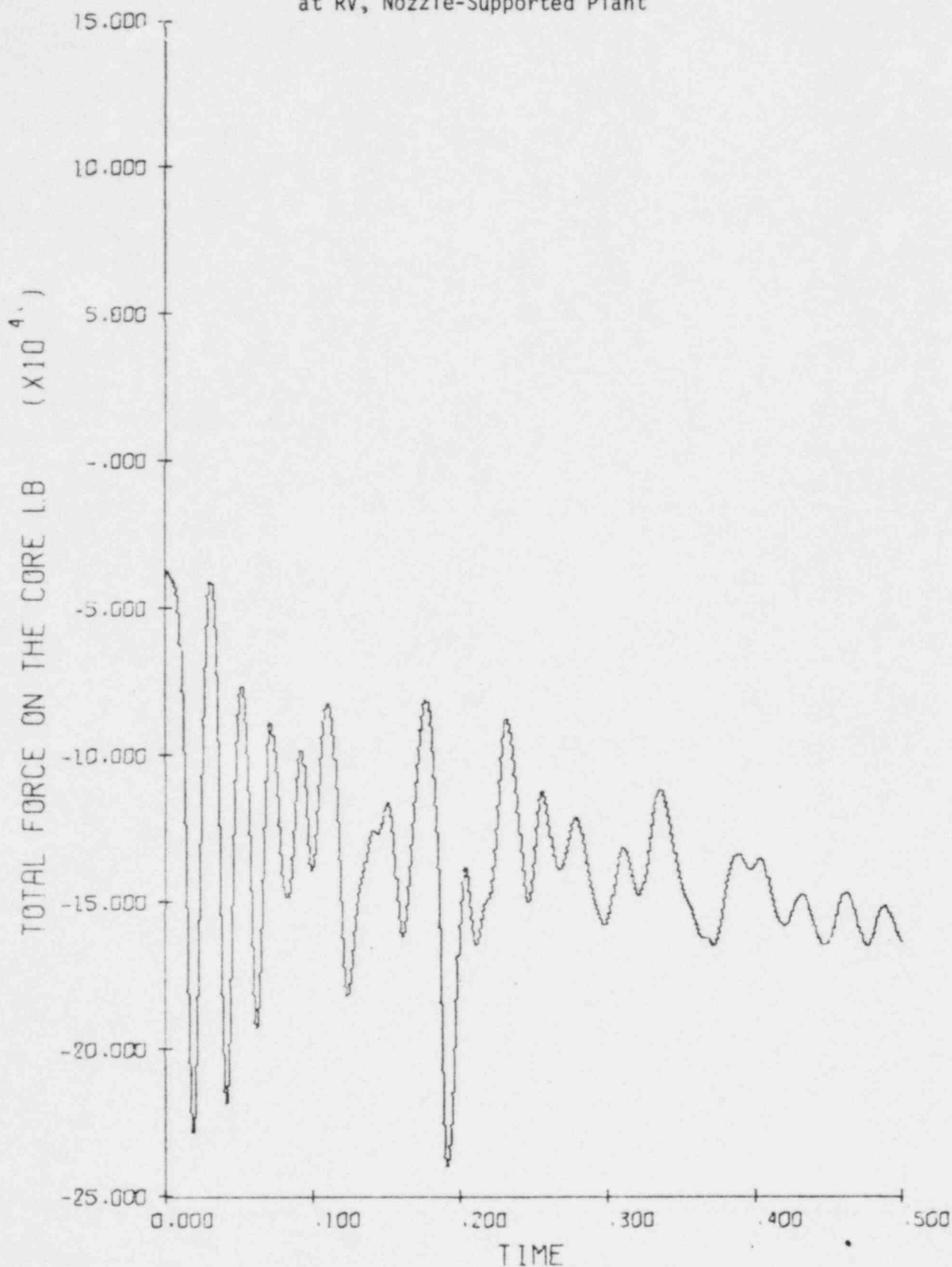
Figure 2.21-7. Total Force on Core - 1.024 Hot Leg Break at RV, Nozzle-Supported Plant



TECO3FO 1.024A HL AT RV

2.21-62

Figure 2.21-8. Total Force on Core — 0.242A Cold Leg Break at RV, Nozzle-Supported Plant



TECO101 .242A CL BREAK AT RV



2.22. In Appendix C, what are the differences between skirt- and nozzle-supported plants that require separate CRAFT2 models?

Response

The major differences between the skirt and nozzle supported plants are as follows:

- The loop configuration.
- The number of vent valves.
- Basic thermal-hydraulic parameters, such as flow, power, pressure drop, etc. (see the response to question 2.10).

2.23. In Appendix C, paragraph 3, reference is made to simulation of four vent valves. However, in the Davis-Besse 1 Safety Analysis Report no mention can be found of any vent valves for that plant. Please clarify.

Response

See section 4.2.2.2, item 8, Internals Vent Valves, page 4-20 in the FSAR.<sup>10</sup>  
Also see section 4.2.2.3, Evaluation of Internal Vent Valves, and Figures 4-3 and 4-3A in the FSAR.<sup>10</sup>

2.24. Identify 177-FA raised- and lowered-loop plants. What is the relationship between these and the skirt- and nozzle-supported-plants? Provide section 4.1 of BAW-10104.

Response

Nozzle-supported is synonymous with raised-loop. Davis Besse 1 is the only B&W 177-FA nozzle-supported plant. All other B&W 177-FA plants listed in section 1 of reference 2 are skirt-supported.

Skirt-supported is synonymous with lowered-loop. For comparison, Figure 11.1-25<sup>2</sup> is an elevation view of a skirt-supported plant, and Figure 11.1-29<sup>2</sup> is an elevation view of the nozzle-supported plant. The basic difference between the 177-FA raised- and lowered-loop plants is the elevation of the steam generator with respect to the reactor vessel inlet and outlet nozzle centerline.

It was agreed at the April 2-3 meeting that section 4.1 of BAW-10104 was not required.

- 2.25. In Appendix C, what is the significance in nodalizing volumes that appear to cross the physical boundaries of the lower incore support plate, the flow distributor plate, and the lower grid in Figure C-9 for the nozzle-supported plant as compared to Figure C-2 for the skirt-supported plant? To what extent does this influence the direct simulation and proper modeling of the flow inertia terms and the location of minimum area junctions for RV internals differential pressure calculations?

Response

There is no significant difference between the reactor vessel lower internals noding scheme shown in Figure C-2 for the skirt-supported plants and that in Figure C-9 for the nozzle-supported plant.

Recall that a nodal pressure is defined at the center of a node and that a flow path proceeds from the center of one node to the center of the next node. Therefore, to predict the proper pressure responses in adjacent nodes, one is required to model all flow perturbations (area changes, loss coefficients, etc.) occurring over the path length. As such, inertial effects for any given flow path are properly accounted for.

For stress analysis the lower internals are considered as comprising two assemblies, the lower grid assembly (lower grid plus the flow distributor plate) and the flow distributor head assembly [incore support plate plus the flow distributor head (defined by the boundary between nodes 141 and 142 in Figure C-2 and between nodes 5 and 9 in Figure C-9)]. Knowing a nodal pressure at the top and bottom surfaces of each of these assemblies allows one to properly characterize the hydraulic forcing functions on the two lower internals assemblies. It is not important that any given plate or assembly be spanned by a single flow path, but rather that the path or paths (in series) account for all flow perturbations induced by the component (plate, assembly, etc.) under consideration.

Both the skirt- and nozzle-supported reactor vessel internals models allow for the proper characterization of the hydraulic forcing functions on the two assemblies that make up the lower internals.

- 2.26. For both skirt- and nozzle-supported plants and for both hot and cold leg RV nozzle breaks, and the 2A cold leg SG compartment break (if available), provide (1) computer listing of the CRAFT2 input for RV internals analysis and (2) the following pressure transients:
- The differential pressure across the RV upper and lower head for volume nodes 141-157 in Figure C-2 and nodes 5-10 in Figure C-9.
  - The differential pressure in the downcomer annulus across volume nodes 97-91, 109-103, and 133-127 for Figure C-2. Provide the same data at the same approximate locations for the nozzle-supported plant.
  - Pressures in the RV for volume nodes 153, 141, 97, 91, and 127 in Figure C-2. Provide the same data at the same approximate locations for the nozzle-supported plant.
  - Differential pressures across the vent valves across volume nodes 169-186 and 162-180 in Figure C-7. Provide the same data at the same approximate locations for the nozzle-supported plant.

Response

Table 2.26-1 lists the production run names associated with the data requested in this question. Table 2.26-2 provides a CRAFT2 reactor vessel internals code input listing for the skirt-supported plants. Table 2.26-3 is the same as Table 2.26-2 except that it is for the nozzle-supported plant.

Copies of microfiche plots for all runs listed in Table 2.26-1 are presented in Table 2.26-4.

Table 2.26-1. Production Run Names

<u>Break Location</u>	<u>Production run name/break area</u>	
	<u>Skirt-supported</u>	<u>Nozzle-supported</u>
Reactor vessel outlet	G15AB29/1.5A G10ABNS/1.0A G06ABN7/0.6A	TEC03F0/1.024A
Reactor vessel inlet	G20ACLK/2.0A	TEC010I/0.242A
Cold leg SG compartment	G2ACGXW/2.0A	TEC2BQR/1.167A

Table 2.26-2. Listing of CRAFT2 Code Inputs for Skirt-Supported Reactor Vessel Internals ΔP Model

LISTING OF INPUT DATA FOR CASE 1

	* 177 OWNERS GROUP GENERIC LOCA LOADS MODEL
	* 2A (8.55 FT) BREAK IN COLD LEG AT REACTOR VESSEL INLET
	* REFERENCE CALCULATION FILE 32-9616-00
	* REFERENCE CORE POWER=2560 MW
	* TOTAL SYSTEM FLOW=131.5+06 LEM/HR
	* BYRON JACKSON REACTOR COOLANT PUMPS
	* MOODY DISCHARGE MODEL, DISCHARGE COEFFICIENT CD=1.21
	*****
	* TIME INFORMATION
*****	101,0.3
*****	1001,0.3,0.0,0.0271
*****	1002,0.,0.,0.05,.0062,0.15,.0116,0.3,.0159,0.5,.020,0.75,.0242,1.00,.027
*****	+1,1.0,1.0+06
*****	1004,10.,10.,10.
	*****
	* PLOT INFORMATION
*****	1500,1
*****	1501,1102,1,0.,2.,.4,1,0.,0.,.6.,0,0,1104,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1106,1,0.,2.,.4,1,0.,0.,.6.,0,0,1108,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1502,1110,1,0.,2.,.4,1,0.,0.,.6.,0,0,1112,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1114,1,0.,2.,.4,1,0.,0.,.6.,0,0,1116,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1503,1118,1,0.,2.,.4,1,0.,0.,.6.,0,0,1120,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1122,1,0.,2.,.4,1,0.,0.,.6.,0,0,1124,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1504,1126,1,0.,2.,.4,1,0.,0.,.6.,0,0,1130,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1134,1,0.,2.,.4,1,0.,0.,.6.,0,0,1138,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1505,1142,1,0.,2.,.4,1,0.,0.,.6.,0,0,1146,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1150,1,0.,2.,.4,1,0.,0.,.6.,0,0,1154,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1506,1158,1,0.,2.,.4,1,0.,0.,.6.,0,0,1162,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1163,1,0.,2.,.4,1,0.,0.,.6.,0,0,1164,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1507,1165,1,0.,2.,.4,1,0.,0.,.6.,0,0,1166,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1167,1,0.,2.,.4,1,0.,0.,.6.,0,0,1169,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1508,1169,1,0.,2.,.4,1,0.,0.,.6.,0,0,1170,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1171,1,0.,2.,.4,1,0.,0.,.6.,0,0,1172,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1509,1173,1,0.,2.,.4,1,0.,0.,.6.,0,0,1174,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1175,1,0.,2.,.4,1,0.,0.,.6.,0,0,1176,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1510,1178,1,0.,2.,.4,1,0.,0.,.6.,0,0,1179,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1180,1,0.,2.,.4,1,0.,0.,.6.,0,0,1181,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1511,1182,1,0.,2.,.4,1,0.,0.,.6.,0,0,1183,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1184,1,0.,2.,.4,1,0.,0.,.6.,0,0,1185,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1512,1186,1,0.,2.,.4,1,0.,0.,.6.,0,0,1187,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1188,1,0.,2.,.4,1,0.,0.,.6.,0,0,1189,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1513,1190,1,0.,2.,.4,1,0.,0.,.6.,0,0,1191,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1192,1,0.,2.,.4,1,0.,0.,.6.,0,0,1193,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1514,1194,1,0.,2.,.4,1,0.,0.,.6.,0,0,1195,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 1196,1,0.,2.,.4,1,0.,0.,.6.,0,0,1197,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1515,1198,1,0.,2.,.4,1,0.,0.,.6.,0,0,1199,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11100,1,0.,2.,.4,1,0.,0.,.6.,0,0,11101,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1516,11102,1,0.,2.,.4,1,0.,0.,.6.,0,0,11103,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11104,1,0.,2.,.4,1,0.,0.,.6.,0,0,11105,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1517,11106,1,0.,2.,.4,1,0.,0.,.6.,0,0,11107,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11108,1,0.,2.,.4,1,0.,0.,.6.,0,0,11109,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1518,11110,1,0.,2.,.4,1,0.,0.,.6.,0,0,11111,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11112,1,0.,2.,.4,1,0.,0.,.6.,0,0,11113,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1519,11114,1,0.,2.,.4,1,0.,0.,.6.,0,0,11115,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11116,1,0.,2.,.4,1,0.,0.,.6.,0,0,11117,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	1520,11118,1,0.,2.,.4,1,0.,0.,.6.,0,0,11119,1,0.,2.,.4,1,0.,0.,.6.,0,0
*****	+ 11120,1,0.,2.,.4,1,0.,0.,.6.,0,0,11121,1,0.,2.,.4,1,0.,0.,.6.,0,0

Table 2.26-2. (Cont'd)

****	1521, 11122, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11123, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11124, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11125, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1522, 11126, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11127, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11128, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11129, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1523, 11130, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11131, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11132, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11133, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1524, 11134, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11135, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11136, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11137, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1525, 11138, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11139, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11140, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11141, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1526, 11142, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11143, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11144, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11145, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1527, 11146, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11147, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11148, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11149, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1528, 11150, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11151, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11152, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11153, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1529, 11154, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11155, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11156, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11157, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1530, 11158, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11159, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11160, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11161, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1531, 11162, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11163, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11164, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11165, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1532, 11166, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11167, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11168, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11169, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1533, 11170, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11171, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
+	11172, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0, 11173, 1.0, 2.4, 1.0, 0.6, 0.0, 0.0
****	1534, 1801, 1.0, 2.4, 1.0, 0.6, 141, 142,
+	1802, 1.0, 2.4, 1.0, 0.6, 142, 143,
****	1535, 1803, 1.0, 2.4, 1.0, 0.6, 143, 144,
+	1804, 1.0, 2.4, 1.0, 0.6, 144, 145,
****	1536, 1805, 1.0, 2.4, 1.0, 0.6, 145, 152,
+	5701, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1537, 25219, 1.0, 2.4, 1.0, 0.6, 0.0, 25220, 1.0, 2.4, 1.0, 0.6, 0.0,
+	25221, 1.0, 2.4, 1.0, 0.6, 0.0, 25222, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1538, 21303, 1.0, 2.4, 1.0, 0.6, 0.0, 21304, 1.0, 2.4, 1.0, 0.6, 0.0,
+	22303, 1.0, 2.4, 1.0, 0.6, 0.0, 22304, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1539, 62303, 1.0, 2.4, 1.0, 0.6, 0.0, 62304, 1.0, 2.4, 1.0, 0.6, 0.0,
+	85303, 1.0, 2.4, 1.0, 0.6, 0.0, 85304, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1540, 11174, 1.0, 2.4, 1.0, 0.6, 0.0, 11175, 1.0, 2.4, 1.0, 0.6, 0.0,
+	11176, 1.0, 2.4, 1.0, 0.6, 0.0, 25299, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1541, 21250, 1.0, 2.4, 1.0, 0.6, 0.0, 21251, 1.0, 2.4, 1.0, 0.6, 0.0,
+	21264, 1.0, 2.4, 1.0, 0.6, 0.0, 21265, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1542, 21266, 1.0, 2.4, 1.0, 0.6, 0.0, 21267, 1.0, 2.4, 1.0, 0.6, 0.0,
+	21229, 1.0, 2.4, 1.0, 0.6, 0.0, 21231, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1543, 21252, 1.0, 2.4, 1.0, 0.6, 0.0, 21253, 1.0, 2.4, 1.0, 0.6, 0.0,
+	21254, 1.0, 2.4, 1.0, 0.6, 0.0, 21255, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1544, 21276, 1.0, 2.4, 1.0, 0.6, 0.0, 21277, 1.0, 2.4, 1.0, 0.6, 0.0,
+	21278, 1.0, 2.4, 1.0, 0.6, 0.0, 21279, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1545, 21280, 1.0, 2.4, 1.0, 0.6, 0.0, 21281, 1.0, 2.4, 1.0, 0.6, 0.0,
+	2172, 1.0, 2.4, 1.0, 0.6, 0.0, 2174, 1.0, 2.4, 1.0, 0.6, 0.0,
****	1601, 1801, 1802, 1803, 1804, 1805, 21303, 21304, 22303, 22304, 62303, 62304
****	1602, 85303, 85304, 5701
****	1701, 10, 10, 10

\*\*\*\*\*  
\* TIME STEP INFORMATION

2001, 1.0-05, 0.050, 50, 1.0-05, 0.15, 1.0-05, 0.3, 4,

\*\*\*\*\*  
\* CONTROL VOLUMES



Table 2.26-3. (Cont'd)

* HOT LEG PIPING VOLUMES						
3001	16.665	4.025	628.465	627.0	625.5	2175.24, 618.726, 0.0
3002	16.665	4.025	628.465	627.0	625.5	2175.24, 618.726, 0.0
3003	6.614	10.396	637.6	628.465	627.404	2169.62, 618.726, 0.0
3004	6.614	10.396	637.6	628.465	627.404	2169.62, 618.726, 0.0
3005	7.069	9.623	647.223	638.0	637.8	2166.28, 618.726, 0.0
3006	7.069	9.623	647.223	638.0	637.8	2166.28, 618.726, 0.0
3007	7.069	9.623	656.847	647.623	647.427	2162.94, 618.726, 0.0
3008	7.069	9.623	656.847	647.623	647.427	2162.94, 618.726, 0.0
3009	7.069	9.623	666.470	657.247	657.047	2159.61, 618.726, 0.0
3010	7.069	9.623	666.470	657.247	657.047	2159.61, 618.726, 0.0
3011	8.190	8.538	673.708	666.870	666.670	2156.02, 618.726, 0.0
3012	8.190	8.538	673.708	666.870	666.670	2156.02, 618.726, 0.0
3013	8.265	8.0	667.408	673.708	667.208	2153.57, 618.726, 0.0
3014	8.265	8.0	667.408	673.708	667.208	2153.57, 618.726, 0.0
*****						
* STEAM GENERATOR VOLUMES						
3015	37.41	7.467	659.941	667.008	659.741	2166.55, 618.726, 0.0
3016	37.41	7.467	659.941	667.008	659.741	2166.55, 618.726, 0.0
3017	26.155	11.25	648.691	659.541	648.491	2166.76, 605.433, 0.0
3018	26.155	11.25	648.691	659.541	648.491	2166.76, 605.433, 0.0
3019	26.155	11.25	637.441	648.291	637.241	2167.17, 592.140, 0.0
3020	26.155	11.25	637.441	648.291	637.241	2167.17, 592.140, 0.0
3021	26.155	11.25	626.191	637.041	625.991	2165.63, 578.846, 0.0
3022	26.155	11.25	626.191	637.041	625.991	2165.63, 578.846, 0.0
3023	26.155	11.25	614.941	625.791	614.741	2165.94, 565.553, 0.0
3024	26.155	11.25	614.941	625.791	614.741	2165.94, 565.553, 0.0
3025	26.155	11.25	603.691	614.541	603.491	2163.45, 552.260, 0.0
3026	26.155	11.25	603.691	614.541	603.491	2163.45, 552.260, 0.0
3027	27.170	4.960	598.905	603.291	598.530	2159.30, 552.260, 0.0
3028	27.170	4.960	598.905	603.291	598.530	2159.30, 552.260, 0.0
3029	27.170	4.960	598.905	603.291	598.530	2159.30, 552.260, 0.0
3030	27.170	4.960	598.905	603.291	598.530	2159.30, 552.260, 0.0
*****						
* COLD LEG PIPING VOLUMES						
3031	9.868	4.667	600.545	598.905	596.078	2153.30, 552.26, 0.0
3032	9.868	4.667	600.545	598.905	596.078	2153.30, 552.26, 0.0
3033	9.868	4.667	600.545	598.905	596.078	2153.30, 552.26, 0.0
3034	9.868	4.667	600.545	598.905	596.078	2153.30, 552.26, 0.0
3035	4.276	9.918	610.463	600.945	600.745	2147.99, 552.26, 0.0
3036	4.276	9.918	610.463	600.945	600.745	2147.99, 552.26, 0.0
3037	4.276	9.918	610.463	600.945	600.745	2147.99, 552.26, 0.0
3038	4.276	9.918	610.463	600.945	600.745	2147.99, 552.26, 0.0
3039	4.276	9.918	620.382	610.863	610.663	2144.39, 552.26, 0.0
3040	4.276	9.918	620.382	610.863	610.663	2144.39, 552.26, 0.0
3041	4.276	9.918	620.382	610.863	610.663	2144.39, 552.26, 0.0
3042	4.276	9.918	620.382	610.863	610.663	2144.39, 552.26, 0.0
3043	4.545	8.975	629.357	620.782	620.582	2141.12, 552.26, 0.0
3044	4.545	8.975	629.357	620.782	620.582	2141.12, 552.26, 0.0
3045	4.545	8.975	629.357	620.782	620.582	2141.12, 552.26, 0.0
3046	4.545	8.975	629.357	620.782	620.582	2141.12, 552.26, 0.0
3047	34.43	2.614	630.5	629.756	629.556	2229.39, 552.26, 0.0
3048	34.43	2.614	630.5	629.756	629.556	2229.39, 552.26, 0.0
3049	34.43	2.614	630.5	629.756	629.556	2229.39, 552.26, 0.0
3050	34.43	2.614	630.5	629.756	629.556	2229.39, 552.26, 0.0
3051	6.859	5.833	627.0	630.5	625.833	2228.23, 552.26, 0.0
3052	6.859	5.833	627.0	630.5	625.833	2228.23, 552.26, 0.0
3053	6.859	5.833	627.0	630.5	625.833	2228.23, 552.26, 0.0

Table 2.26-2. (Cont'd)

****	3054	6.859	5.833	627.0	630.5	625.833	2228.23	552.26	0.0
****	3055	12.923	2.333	627.0	627.0	625.833	2227.70	552.26	0.0
****	3056	12.923	2.333	627.0	627.0	625.833	2227.70	552.26	0.0
****	3057	12.923	2.333	627.0	627.0	625.833	2227.70	552.26	0.0
****	3058	12.923	2.333	627.0	627.0	625.833	2227.70	552.26	0.0
****	3059	10.973	2.333	627.0	627.0	625.833	2226.11	552.26	0.0
****	3060	10.973	2.333	627.0	627.0	625.833	2226.11	552.26	0.0
****	3061	10.973	2.333	627.0	627.0	625.833	2226.11	552.26	0.0
****	3062	10.973	2.333	627.0	627.0	625.833	2226.11	552.26	0.0
*****									
* BROKEN COLD LEG RV INLET NOZZLE VOLUMES									
****	3063	1.632	2.333	627.0	627.0	625.833	2225.92	552.26	0.0
****	3064	1.632	2.333	627.0	627.0	625.833	2225.82	552.26	0.0
****	3065	1.498	2.597	627.0	627.0	625.65	2225.76	552.26	0.0
****	3066	1.711	3.107	627.0	627.0	625.447	2225.83	552.26	0.0
*****									
****	3067	5.374	3.107	627.0	627.0	625.447	2225.84	552.26	0.0
****	3068	5.374	3.107	627.0	627.0	625.447	2225.84	552.26	0.0
****	3069	5.374	3.107	627.0	627.0	625.447	2225.84	552.26	0.0
****	3070	8.918	4.183	627.0	627.0	624.908	2175.73	618.726	0.0
*****									
* BROKEN HOT LEG RV OUTLET NOZZLE VOLUMES									
****	3071	2.009	4.183	627.0	627.0	624.908	2186.78	618.726	0.0
****	3072	1.822	3.826	627.0	627.0	625.087	2182.93	618.726	0.0
****	3073	1.636	3.469	627.0	627.0	625.266	2179.04	618.726	0.0
****	3074	1.532	3.112	627.0	627.0	625.444	2175.89	618.726	0.0
****	3075	1.914	3.0	627.0	627.0	625.5	2175.89	618.726	0.0
****	3076	1.914	3.0	627.0	627.0	625.5	2175.89	618.726	0.0
*****									
* PRESSURIZER CONTROL VOLUME									
****	3077	36.35	42.571	623.244	623.244	623.044	2163.69	0.0	22.255
*****									
* DOWNCOMER VOLUMES									
****	3078	1.807	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3079	0.476	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3080	2.484	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3081	0.476	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3082	1.807	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3083	2.723	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3084	1.807	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3085	0.476	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3086	2.484	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3087	0.476	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3088	1.807	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3089	2.723	3.362	631.16	631.16	629.474	2230.11	552.26	0.0
****	3090	2.810	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3091	0.593	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3092	4.211	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3093	0.593	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3094	2.810	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3095	5.586	4.948	627.0	628.884	624.136	2188.75	618.726	0.0
****	3096	2.810	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3097	0.593	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3098	4.211	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3099	0.593	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3100	2.810	4.948	627.0	627.0	624.526	2231.20	552.26	0.0
****	3101	5.586	4.948	627.0	628.884	624.136	2188.75	618.726	0.0
****	3102	3.230	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3103	0.682	4.948	622.052	622.052	619.578	2226.30	552.26	0.0

Table 2.26-2. (Cont'd)

****	3104	4.816	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3105	0.682	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3106	3.206	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3107	3.900	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3108	3.230	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3109	0.682	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3110	4.840	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3111	0.682	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3112	3.206	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3113	3.900	4.948	622.052	622.052	619.578	2226.30	552.26	0.0
****	3114	3.348	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3115	0.707	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3116	4.936	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3117	0.707	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3118	3.266	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3119	4.046	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3120	3.348	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3121	0.707	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3122	5.018	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3123	0.707	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3124	3.266	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3125	4.046	4.948	617.104	617.104	614.63	2228.03	552.26	0.0
****	3126	3.348	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3127	0.707	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3128	5.018	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3129	0.707	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3130	3.348	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3131	4.046	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3132	3.348	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3133	0.707	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3134	5.018	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3135	0.707	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3136	3.348	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
****	3137	4.046	7.328	610.966	610.966	607.302	2229.55	552.26	0.0
*****									
* THERMAL SHIELD ANNULUS VOLUMES									
****	3138	3.186	3.922	619.778	623.30	619.578	2227.30	552.26	0.0
****	3139	3.186	4.948	614.830	619.378	614.63	2228.56	552.26	0.0
****	3140	3.186	5.005	609.825	614.43	609.625	2230.15	552.26	0.0
*****									
* LOWER HEAD THROUGH LOWER GRID VOLUMES									
****	3141	55.756	4.885	605.144	607.102	602.417	2229.65	552.26	0.0
****	3142	47.243	1.792	606.936	605.544	605.344	2225.37	552.26	0.0
****	3143	64.833	1.740	608.676	607.336	607.136	2222.33	552.26	0.0
****	3144	86.102	0.531	609.207	609.076	608.876	2221.06	552.26	0.0
****	3145	55.407	1.130	610.337	609.607	609.407	2219.65	552.26	0.0
*****									
* CORE VOLUMES									
****	3146	59.5	4.0	614.337	610.737	610.537	2215.51	563.338	0.0
****	3147	59.5	4.0	618.337	614.737	614.537	2209.29	585.492	0.0
****	3148	59.5	4.0	622.337	618.737	618.537	2205.32	607.648	0.0
*****									
* INNER CORE BYPASS VOLUMES									
****	3149	14.55	4.535	614.337	610.202	610.002	2213.98	552.26	0.0
****	3150	14.55	4.0	618.337	614.737	614.537	2209.55	552.26	0.0
****	3151	14.55	4.798	622.936	618.737	618.537	2204.98	552.26	0.0
*****									
* UPPER PLENUM VOLUMES									

Table 2.26-2. (Cont'd)

*****	3152,72.672,2.344,624.681,622.740,622.537,2204.05,618.726,0.0
*****	3153,52.972,4.812,629.493,625.091,624.891,2201.45,618.726,0.0
*****	3154,52.972,4.282,633.775,629.893,629.693,2200.09,618.726,0.0
*****	3155,20.620,4.812,629.493,625.091,624.881,2201.17,611.726,0.0
*****	3156,20.620,6.053,635.546,629.893,629.693,2199.04,611.726,0.0
*****	3157,67.275,5.734,634.175,635.546,633.975,2197.49,611.726,0.0
*****	3158,8.399,3.579,634.175,635.764,633.975,2198.04,618.726,0.0
*****	3159,8.399,3.579,634.175,635.764,633.975,2198.04,618.726,0.0
*****	3160,8.399,3.579,634.175,635.764,633.975,2198.04,618.726,0.0
*****	3161,8.399,3.579,634.175,635.764,633.975,2198.04,618.726,0.0
*****	
* OUTLET ANNULUS VOLUMES	
*****	3162,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3163,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3164,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3165,2.793,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3166,5.586,4.948,627.0,628.884,624.136,2188.75,618.726,0.0
*****	3167,5.586,4.948,627.0,628.884,624.136,2188.75,618.726,0.0
*****	3168,2.793,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3169,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3170,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3171,1.862,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3172,2.793,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	3173,2.793,4.083,629.284,632.967,629.084,2198.85,618.726,0.0
*****	
* CONTAINMENT NODE	
*****	3174,2.205+04,100.0,627.0,627.0,577.0,14.7,0.0,0.001
*****	
* CONTAINMENT NODE	
*****	3801,174,303,304
*****	
* FLOW PATHS	
* CORE PATHS	
*****	4001,1.145,146,3.573+04,0.052,49.64,2.356,0.0,0.0,0.0,1.238,0.048
*****	4002,1.146,147,3.573+04,0.079,49.64,0.0,-3.414-09,0.0,0.0,2.425,0.048
*****	4003,1.147,148,3.573+04,0.079,49.64,0.0,-4.024-09,0.0,0.0,2.405,0.048
*****	4004,1.148,152,3.573+04,0.065,49.64,2.744,0.0,0.0,0.0,1.0-04,0.048
*****	
* HOT LEG PIPING PATHS	
*****	4005,8,5,7,1.826+04,1.361,7.069,0.0,0.0,0.0,0.0,0.0274,3.0
*****	4006,8,6,8,1.826+04,1.361,7.069,0.0,0.0,0.0,0.0,0.0274,3.0
*****	4007,8,7,9,1.826+04,1.361,7.069,0.0,0.0,0.0,0.0,0.0274,3.0
*****	4008,8,8,10,1.826+04,1.361,7.069,0.0,0.0,0.0,0.0,0.0274,3.0
*****	4009,8,9,11,1.826+04,1.381,7.069,0.0436,0.0,0.0,0.0,0.0,3.0
*****	4010,8,10,12,1.826+04,1.381,7.069,0.0436,0.0,0.0,0.0,0.0,0.0195,3.0
*****	4011,8,11,13,1.826+04,1.361,7.069,0.1442,0.0,0.0,0.0,1.0-04,3.0
*****	4012,8,12,14,1.826+04,1.361,7.069,0.1442,0.0,0.0,0.0,1.0-04,3.0
*****	4013,8,13,15,1.826+04,1.058,7.069,0.1526,1.705-06,0.0,0.0,0.0086,3.0
*****	4014,8,14,16,1.826+04,1.058,7.069,0.1526,1.705-06,0.0,0.0,0.0086,3.0
*****	
* STEAM GENERATOR PATHS	
*****	4015,8,15,17,1.826+04,0.247,26.155,1.242,4.323-07,0.0,0.0,1.7185,0.046
*****	4016,8,16,18,1.826+04,0.247,26.155,1.242,4.323-07,0.0,0.0,1.7185,0.046
*****	4017,8,17,19,1.826+04,0.430,26.155,0.0,0.0,0.0,0.0,3.437,0.046
*****	4018,8,18,20,1.826+04,0.430,26.155,0.0,0.0,0.0,0.0,3.437,0.046
*****	4019,8,19,21,1.826+04,0.430,26.155,0.0,0.0,0.0,0.0,3.437,0.046
*****	4020,8,20,22,1.826+04,0.430,26.155,0.0,0.0,0.0,0.0,3.437,0.046
*****	4021,8,21,23,1.826+04,0.430,26.155,0.0,0.0,0.0,0.0,3.437,0.046

Table 2.26-2. (Cont'd)

****	4022	8,22,24	1.826+04	0.430	26.155	0.0	0.0	0.0	0.0	0.0	0.0	3.437	0.046
****	4023	8,23,25	1.826+04	0.430	26.155	0.0	0.0	0.0	0.0	0.0	0.0	3.437	0.046
****	4024	8,24,26	1.826+04	0.430	26.155	0.0	0.0	0.0	0.0	0.0	0.0	3.437	0.046
****	4025	8,25,27	9132.0	0.5005	13.078	0.5804	-2.9034-06	0.0	0.0	0.0	0.0	1.7188	0.046
****	4026	8,26,28	9132.0	0.5005	13.078	0.5804	-2.9034-06	0.0	0.0	0.0	0.0	1.7188	0.046
****	4027	8,25,29	9132.0	0.5005	13.078	0.5804	-2.9034-06	0.0	0.0	0.0	0.0	1.7188	0.046
****	4028	8,26,30	9132.0	0.5005	13.078	0.5804	-2.9034-06	0.0	0.0	0.0	0.0	1.7188	0.046
*****													
* COLD LEG PIPING PATHS													
****	4029	8,27,31	9132.0	1.8102	4.276	0.146	-2.9084-06	0.0	0.0	0.0	0.0	0.0138	2.333
****	4030	8,28,32	9132.0	1.8102	4.276	0.146	-2.9084-06	0.0	0.0	0.0	0.0	0.0138	2.333
****	4031	8,29,33	9132.0	1.8102	4.276	0.146	-2.9084-06	0.0	0.0	0.0	0.0	0.0138	2.333
****	4032	8,30,34	9132.0	1.8102	4.276	0.146	-2.9084-06	0.0	0.0	0.0	0.0	0.0138	2.333
****	4033	8,31,35	9132.0	2.419	4.276	0.1804	0.0	0.0	0.0	0.0	0.0	0.0188	2.333
****	4034	8,32,36	9132.0	2.419	4.276	0.1804	0.0	0.0	0.0	0.0	0.0	0.0188	2.333
****	4035	8,33,37	9132.0	2.419	4.276	0.1804	0.0	0.0	0.0	0.0	0.0	0.0188	2.333
****	4036	8,34,38	9132.0	2.419	4.276	0.1804	0.0	0.0	0.0	0.0	0.0	0.0188	2.333
****	4037	8,35,39	9132.0	2.320	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0368	2.333
****	4038	8,36,40	9132.0	2.320	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0368	2.333
****	4039	8,37,41	9132.0	2.320	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0368	2.333
****	4040	8,38,42	9132.0	2.320	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0368	2.333
****	4041	8,39,43	9132.0	2.209	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0352	2.333
****	4042	8,40,44	9132.0	2.209	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0352	2.333
****	4043	8,41,45	9132.0	2.209	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0352	2.333
****	4044	8,42,46	9132.0	2.209	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.0352	2.333
*****													
* PUMP PATHS													
****	4045	2,43,47	9132.0	0.957	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.012	2.333
****	4046	2,44,48	9132.0	0.957	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.012	2.333
****	4047	2,45,49	9132.0	0.957	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.012	2.333
****	4048	2,46,50	9132.0	0.957	4.276	0.0	0.0	0.0	0.0	0.0	0.0	0.012	2.333
*****													
* COLD LEG PIPING PATHS													
****	4049	8,47,51	9132.0	1.680	4.276	0.118	0.0	0.0	0.0	0.0	0.0	7.091-03	2.333
****	4050	8,48,52	9132.0	1.680	4.276	0.118	0.0	0.0	0.0	0.0	0.0	7.091-03	2.333
****	4051	8,49,53	9132.0	1.680	4.276	0.118	0.0	0.0	0.0	0.0	0.0	7.091-03	2.333
****	4052	8,50,54	9132.0	1.680	4.276	0.118	0.0	0.0	0.0	0.0	0.0	7.091-03	2.333
****	4053	8,51,55	9132.0	1.919	4.276	0.1377	0.0	0.0	0.0	0.0	0.0	0.019	2.333
****	4054	8,52,56	9132.0	1.919	4.276	0.1377	0.0	0.0	0.0	0.0	0.0	0.019	2.333
****	4055	8,53,57	9132.0	1.919	4.276	0.1377	0.0	0.0	0.0	0.0	0.0	0.019	2.333
****	4056	8,54,58	9132.0	1.919	4.276	0.1377	0.0	0.0	0.0	0.0	0.0	0.019	2.333
****	4057	8,55,59	9132.0	1.524	4.276	0.1412	0.0	0.0	0.0	0.0	0.0	0.0097	2.333
****	4058	8,56,60	9132.0	1.524	4.276	0.1412	0.0	0.0	0.0	0.0	0.0	0.0097	2.333
****	4059	8,57,61	9132.0	1.524	4.276	0.1412	0.0	0.0	0.0	0.0	0.0	0.0097	2.333
****	4060	8,58,62	9132.0	1.524	4.276	0.1412	0.0	0.0	0.0	0.0	0.0	0.0097	2.333
****	4061	10,62,63	9132.0	0.804	4.276	0.0	-2.648-08	0.0	0.0	0.0	0.0	0.0141	2.333,1.0
****	4062	8,60,68	9132.0	1.093	4.276	0.0	-9.999-08	0.0	0.0	0.0	0.0	0.0190	2.333
****	4063	8,61,67	9132.0	1.093	4.276	0.0	-9.999-08	0.0	0.0	0.0	0.0	0.0190	2.333
****	4064	8,59,69	9132.0	1.093	4.276	0.0	-9.999-08	0.0	0.0	0.0	0.0	0.0190	2.333
****	4065	8,63,64	9132.0	0.202	4.276	0.0	-5.139-08	0.0	0.0	0.0	0.0	3.473-03	2.333
****	4066	8,68,93	9132.0	0.385	5.102	0.1894	3.947-06	0.0	0.0	0.0	0.0	1.0-03	2.333
****	4067	8,64,65	9132.0	0.192	4.276	0.0	-3.407-08	0.0	0.0	0.0	0.0	2.299-03	2.333
****	4068	8,69,91	9132.0	0.385	5.102	0.1894	3.947-06	0.0	0.0	0.0	0.0	1.0-03	2.333
****	4069	8,65,66	9132.0	0.144	5.695	0.1296	2.186-06	0.0	0.0	0.0	0.0	1.0-04	2.333
****	4070	8,66,97	9132.0	0.135	7.113	0.164	1.673-06	0.0	0.0	0.0	0.0	1.0-03	3.009
****	4071	8,67,99	9132.0	0.385	5.102	0.1894	3.947-06	0.0	0.0	0.0	0.0	1.0-03	2.333
****	4072	8,101,70	1.826+04	0.223	10.845	0.122	-1.557-06	0.0	0.0	0.0	0.0	1.0-03	3.721
****	4073	8,70,1	1.826+04	0.975	7.07	0.0	-4.442-08	0.0	0.0	0.0	0.0	0.0212	3.0

Table 2.26-2. (Cont'd)

****	4074	8,95,71	1.826+04	0.083	13.183	0.027	-2.351-07	0.0	0.0	0.0	1.0-03	4.098
****	4075	8,71,72	1.826+04	0.059	11.523	0.0416	-4.701-07	0.0	0.0	0.0	1.0-03	3.834
****	4076	8,72,73	1.826+04	0.070	9.432	0.028	-4.701-07	0.0	0.0	0.0	1.0-03	3.480
****	4077	8,73,74	1.826+04	0.003	8.234	0.0178	-3.818-07	0.0	0.0	0.0	1.0-03	3.24
****	4078	8,74,75	1.826+04	0.105	7.059	0.000	0.0	0.0	0.0	0.0	0.0	3.0
****	4079	8,75,76	1.826+04	0.115	7.069	0.000	0.0	0.0	0.0	0.0	0.0	3.0
****	4080	8,76,2	1.826+04	0.729	7.069	0.000	0.0	0.0	0.0	0.0	0.0	3.0
****	4081	5,77,4	0.0	142.76	0.4176	0.000	0.0	0.0	0.0	0.0	0.8145	0.729
*****												
* DOWNCOMER FLOW PATHS												
****	4082	8,73,79	0.00	1.611	1.871	0.000	0.0	0.0	0.0	0.0	0.0	2.479 1.00
****	4083	8,79,80	0.00	2.485	1.871	0.000	0.0	0.0	0.0	0.0	0.0	3.500 1.00
****	4084	8,80,81	0.00	2.485	1.871	0.000	0.0	0.0	0.0	0.0	0.0	3.500 1.00
****	4085	8,81,82	0.00	1.611	1.871	0.000	0.0	0.0	0.0	0.0	0.0	2.479 1.00
****	4086	8,82,83	0.00	2.702	1.871	0.000	0.0	0.0	0.0	0.0	0.0	4.521 1.00
****	4087	8,83,84	0.00	2.702	1.871	0.000	0.0	0.0	0.0	0.0	0.0	4.521 1.00
****	4088	8,84,85	0.00	1.611	1.871	0.000	0.0	0.0	0.0	0.0	0.0	2.479 1.00
****	4089	8,85,86	0.00	2.485	1.871	0.000	0.0	0.0	0.0	0.0	0.0	3.500 1.00
****	4090	8,86,87	0.00	2.485	1.871	0.000	0.0	0.0	0.0	0.0	0.0	3.500 1.00
****	4091	8,87,88	0.00	1.611	1.871	0.000	0.0	0.0	0.0	0.0	0.0	2.479 1.00
****	4092	8,88,89	0.00	2.702	1.871	0.000	0.0	0.0	0.0	0.0	0.0	4.521 1.00
****	4093	8,89,90	0.00	2.702	1.871	0.000	0.0	0.0	0.0	0.0	0.0	4.521 1.00
****	4094	8,91,90	913.20	0.723	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.024 1.385
****	4095	8,91,92	913.20	1.021	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.035 1.385
****	4096	8,93,92	913.20	1.021	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.035 1.385
****	4097	8,93,94	913.20	0.723	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.024 1.385
****	4098	8,97,96	913.20	0.723	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.024 1.385
****	4099	8,97,98	913.20	1.021	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.035 1.385
****	4100	8,99,98	913.20	1.021	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.035 1.385
****	4101	8,99,100	913.20	0.723	3.428	0.000	0.0	0.0	0.0	0.0	0.0	0.024 1.385
****	4102	8,103,102	4794.14	0.634	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.021 1.580
****	4103	8,103,104	1750.45	0.895	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.030 1.580
****	4104	8,105,104	1750.48	0.910	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.030 1.580
****	4105	8,105,106	4794.14	0.634	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.021 1.580
****	4106	8,106,107	2148.88	1.172	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.039 1.580
****	4107	8,108,107	2148.88	1.156	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.039 1.580
****	4108	8,109,103	4794.14	0.634	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.021 1.580
****	4109	8,109,110	1750.45	0.895	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.030 1.580
****	4110	8,111,110	1750.48	0.895	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.030 1.580
****	4111	8,111,112	4794.14	0.634	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.021 1.580
****	4112	8,112,113	2148.88	1.172	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.039 1.580
****	4113	8,102,113	2148.88	1.156	3.910	0.000	0.0	0.0	0.0	0.0	0.0	0.039 1.580
****	4114	8,114,115	0.0	0.613	4.046	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635
****	4115	8,115,116	0.0	0.865	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4116	8,116,117	0.0	0.987	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4117	8,117,118	0.0	0.613	4.046	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635
****	4118	8,118,119	0.0	1.240	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.765 1.635
****	4119	8,119,120	0.0	1.117	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.765 1.635
****	4120	8,120,121	0.0	0.613	4.046	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635
****	4121	8,121,122	0.0	0.865	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4122	8,122,123	0.0	0.865	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4123	8,123,124	0.0	0.613	4.046	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635
****	4124	8,124,125	0.0	1.240	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.765 1.635
****	4125	8,125,114	0.0	1.117	4.046	0.000	0.0	0.0	0.0	0.0	0.0	2.765 1.635
****	4126	8,126,127	0.0	0.414	5.992	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635
****	4127	8,127,128	0.0	0.584	5.992	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4128	8,128,129	0.0	0.584	5.992	0.000	0.0	0.0	0.0	0.0	0.0	2.141 1.635
****	4129	8,129,130	0.0	0.414	5.992	0.000	0.0	0.0	0.0	0.0	0.0	1.516 1.635

Table 2.26-2. (Cont'd)

4130	8	130	131	0.0	0.755	5.992	0.0	0.0	0.0	0.0	0.0	2.765	1.635	
4131	8	131	132	0.0	0.755	5.992	0.0	0.0	0.0	0.0	0.0	2.765	1.635	
4132	8	132	133	0.0	0.414	5.992	0.0	0.0	0.0	0.0	0.0	1.516	1.635	
4133	8	133	134	0.0	0.584	5.992	0.0	0.0	0.0	0.0	0.0	2.141	1.635	
4134	8	134	135	0.0	0.584	5.992	0.0	0.0	0.0	0.0	0.0	2.141	1.635	
4135	8	135	136	0.0	0.414	5.992	0.0	0.0	0.0	0.0	0.0	1.516	1.635	
4136	8	136	137	0.0	0.755	5.992	0.0	0.0	0.0	0.0	0.0	2.765	1.635	
4137	8	137	126	0.0	0.755	5.992	0.0	0.0	0.0	0.0	0.0	2.765	1.635	
4138	8	90	78	0.0	1.465	2.836	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4139	8	91	79	0.0	6.941	0.599	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4140	8	92	80	0.0	0.978	4.250	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4141	8	93	81	0.0	6.941	0.599	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4142	8	94	82	0.0	1.465	2.836	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4143	8	96	84	0.0	1.465	2.836	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4144	8	97	85	0.0	6.941	0.599	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4145	8	98	86	0.0	0.978	4.250	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4146	8	99	87	0.0	6.941	0.599	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4147	8	100	88	0.0	1.465	2.836	0.0	0.0	0.0	0.0	0.0	2.999	1.385	
4148	8	90	102	913.20	1.661	2.979	33.05	3.732	-05	0.0	0.0	1.0-03	1.456	
4149	8	91	103	7305.60	7.868	0.629	0.023	5.831	-07	0.0	0.0	1.0-03	1.456	
4150	8	92	104	1826.40	1.108	4.465	18.56	9.33	-06	0.0	0.0	1.0-03	1.456	
4151	8	93	105	7305.60	7.868	0.629	0.023	5.831	-07	0.0	0.0	1.0-03	1.456	
4152	8	94	106	913.20	1.661	2.979	33.05	3.732	-05	0.0	0.0	1.0-03	1.456	
4153	8	96	108	913.20	1.661	2.979	33.05	3.732	-05	0.0	0.0	1.0-03	1.456	
4154	8	97	109	7305.60	7.868	0.629	0.023	5.831	-07	0.0	0.0	1.0-03	1.456	
4155	8	98	110	1826.40	1.108	4.465	18.56	9.33	-06	0.0	0.0	1.0-03	1.456	
4156	8	99	111	7305.60	7.868	0.629	0.023	5.831	-07	0.0	0.0	1.0-03	1.456	
4157	8	100	112	913.20	1.661	2.979	33.05	3.732	-05	0.0	0.0	1.0-03	1.456	
4158	8	102	114	3546.28	1.478	3.348	0.0	0.8	028	-07	0.0	0.0	0.0296	1.635
4159	8	103	115	748.87	7.002	0.707	0.0	1.713	-05	0.0	0.0	0.028	1.635	
4160	8	104	116	5315.19	0.999	5.018	0.029	3.54	-07	0.0	0.0	1.0-03	1.635	
4161	8	105	117	748.87	7.002	0.707	0.0	1.713	-05	0.0	0.0	0.028	1.635	
4162	8	106	118	3546.28	1.507	3.348	0.029	8.028	-07	0.0	0.0	1.0-03	1.635	
4163	8	107	119	4285.62	1.223	4.046	0.0	5.518	-07	0.0	0.0	0.0297	1.635	
4164	8	108	120	3546.28	1.478	3.348	0.0	0.8	028	-07	0.0	0.0	0.0296	1.635
4165	8	109	121	748.87	7.002	0.707	0.0	1.713	-05	0.0	0.0	0.028	1.635	
4166	8	110	122	5315.19	0.986	5.018	0.0	3.54	-07	0.0	0.0	0.029	1.635	
4167	8	111	123	748.87	7.002	0.707	0.0	1.713	-05	0.0	0.0	0.028	1.635	
4168	8	112	124	3546.28	1.507	3.348	0.029	8.028	-07	0.0	0.0	1.0-03	1.635	
4169	8	113	125	4285.62	1.223	4.046	0.0	5.518	-07	0.0	0.0	0.0297	1.635	
4170	8	114	126	3546.28	1.833	3.348	0.0	0.0	0.0	0.0	0.0	0.0298	1.635	
4171	8	115	127	748.87	8.682	0.707	0.0	0.0	0.0	0.0	0.0	0.029	1.635	
4172	8	116	128	5315.19	1.232	5.018	0.028	0.0	0.0	0.0	0.0	1.0-03	1.635	
4173	8	117	129	748.87	8.682	0.707	0.0	0.0	0.0	0.0	0.0	0.029	1.635	
4174	8	118	130	3546.28	1.853	3.348	0.030	0.0	0.0	0.0	0.0	1.0-03	1.635	
4175	8	119	131	4285.62	1.517	4.046	0.0	0.0	0.0	0.0	0.0	0.030	1.635	
4176	8	120	132	3546.28	1.833	3.348	0.0	0.0	0.0	0.0	0.0	0.0298	1.635	
4177	8	121	133	748.87	8.682	0.707	0.0	0.0	0.0	0.0	0.0	0.029	1.635	
4178	8	122	134	5315.19	1.223	5.018	0.0	0.0	0.0	0.0	0.0	0.030	1.635	
4179	8	123	135	748.87	8.682	0.707	0.0	0.0	0.0	0.0	0.0	0.029	1.635	
4180	8	124	136	3546.28	1.853	3.348	0.030	0.0	0.0	0.0	0.0	1.0-03	1.635	
4181	8	125	137	4285.62	1.517	4.046	0.0	0.0	0.0	0.0	0.0	0.030	1.635	
4182	8	126	141	3545.55	1.547	3.348	1.807	8.102	-06	0.0	0.0	1.0-04	1.0	
4183	8	127	141	748.87	7.326	0.707	1.753	1.729	-04	0.0	0.0	1.0-04	1.0	
4184	8	128	141	5314.46	1.032	5.018	1.830	3.576	-06	0.0	0.0	1.0-04	1.0	
4185	8	129	141	748.87	7.326	0.707	1.753	1.729	-04	0.0	0.0	1.0-04	1.0	
4186	8	130	141	3545.55	1.547	3.348	1.807	8.102	-06	0.0	0.0	1.0-04	1.0	
4187	8	131	141	4285.62	1.280	4.046	1.826	5.569	-06	0.0	0.0	1.0-04	1.0	

Table 2.26-2. (Cont'd)

****	4188	8	132	141	3545.55	1.547	3.348	1.807	8.102-06	0.0	0.0	0.0	1.0-04	1.0
****	4189	8	133	141	748.87	7.326	0.707	1.753	1.729-04	0.0	0.0	0.0	1.0-04	1.0
****	4190	8	134	141	534.46	1.032	5.018	1.830	3.576-06	0.0	0.0	0.0	1.0-04	1.0
****	4191	8	135	141	748.87	7.326	0.707	1.753	1.729-04	0.0	0.0	0.0	1.0-04	1.0
****	4192	8	136	141	3545.55	1.547	3.348	1.807	8.102-06	0.0	0.0	0.0	1.0-04	1.0
****	4193	8	137	141	4265.62	1.280	4.046	1.826	5.569-06	0.0	0.0	0.0	1.0-04	1.0
*****														
* THERMAL SHIELD ANNULUS PATHS														
****	4194	8	102	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4195	8	103	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4196	8	104	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4197	8	105	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4198	8	106	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4199	8	107	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4200	8	108	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4201	8	109	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4202	8	110	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4203	8	111	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4204	8	112	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4205	8	113	138	12.14	0.616	3.185	88.99	-2.2104-04	0.0	0.0	0.0	1.0-04	0.1667
****	4206	8	138	139	145.523	1.392	3.185	0.0	-2.851-06	0.0	0.0	0.0	1.031	0.1667
****	4207	8	139	140	145.528	1.554	3.185	0.0	0.0	0.0	0.0	0.0	0.824	0.1667
****	4208	8	126	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4209	8	128	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4210	8	130	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4211	8	132	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4212	8	134	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4213	8	136	140	0.736	119.07	0.0014	1.2024	0.0	0.0	0.0	0.0	1.0-04	0.042
****	4214	8	140	149	149.944	1.812	0.092	0.762	0.0	0.0	0.0	0.0	1.856	0.042
*****														
* LOWER HEAD THROUGH LOWER GRID FLOW PATHS														
****	4215	8	141	142	3.633+04	0.043	30.965	1.366	1.020-09	0.0	0.0	0.0	1.0-04	1.0
****	4216	8	142	143	3.633+04	0.035	35.46	0.609	-3.543-08	0.0	0.0	0.0	1.0-04	1.0
****	4217	8	143	144	3.633+04	0.015	44.88	1.385	3.5740-08	0.0	0.0	0.0	1.0-04	1.0
****	4218	8	144	145	3.633+04	0.012	45.650	0.134	-0.30-07	0.0	0.0	0.0	1.0-04	1.0
*****														
* INNER CORE BYPASS FLOW PATHS														
****	4219	8	145	149	651.670	0.718	14.55	944.42	1.535-06	0.0	0.0	0.0	1.0-04	4.30
****	4220	8	149	150	001.614	0.757	14.55	448.87	1.012-06	0.0	0.0	0.0	1.0-04	1.0
****	4221	8	150	151	801.614	0.564	14.55	429.06	0.0	0.0	0.0	0.0	1.0-04	1.002
****	4222	8	151	152	001.614	0.271	14.55	460.706	4.336-07	0.0	0.0	0.0	1.0-03	4.30
*****														
* UPPER PLENUM FLOW PATHS														
****	4223	8	168	95	2559.52	1.500	3.010	6.586	0.0	0.0	0.0	0.0	1.0-04	1.333
****	4224	8	171	167	3383.84	2.252	2.005	1.672	0.0	0.0	0.0	0.0	1.0-04	1.333
****	4225	8	170	167	3383.84	2.252	2.005	1.672	0.0	0.0	0.0	0.0	1.0-04	1.333
****	4226	8	149	146	0.0	0.076	0.825	2.5	0.0	0.0	0.0	0.0	0.544	0.115
****	4227	8	150	147	0.0	0.095	0.66	2.5	0.0	0.0	0.0	0.0	0.543	0.115
****	4228	8	151	148	0.0	0.038	1.65	2.5	0.0	0.0	0.0	0.0	0.543	0.115
****	4229	8	152	153	3.358+04	0.068	57.68	0.618	0.0	0.0	0.0	0.0	1.0-04	1.349
****	4230	8	153	154	2.759+04	0.076	57.68	0.0	0.0	0.0	0.0	0.0	0.151	1.349
****	4231	8	152	155	2951.44	0.164	23.01	19.72	0.0	0.0	0.0	0.0	1.0-03	0.665
****	4232	8	155	156	2951.44	0.256	21.19	0.0	0.0	0.0	0.0	0.0	0.674	0.665
****	4233	8	156	157	2951.44	0.167	21.19	7.674	0.0	0.0	0.0	0.0	1.0-03	0.665
****	4234	8	157	158	737.86	0.120	30.107	0.0	0.0	0.0	0.0	0.0	1.0-03	1.0
****	4235	8	157	159	737.86	0.120	30.107	0.0	0.0	0.0	0.0	0.0	1.0-03	1.0
****	4236	8	157	160	737.86	0.120	30.107	0.0	0.0	0.0	0.0	0.0	1.0-03	1.0
****	4237	8	157	161	737.86	0.120	30.107	0.0	0.0	0.0	0.0	0.0	1.0-03	1.0



Table 2.26-2. (Cont'd)

4235	8	158	162	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4239	8	158	163	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4240	8	153	164	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4241	8	159	165	1584.93	1.743	0.876	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4242	8	154	158	2432.5	2.424	0.860	0.048	0.0	0.0	0.0	0.0	1.0-04	1.0
4243	8	154	159	2432.5	2.424	0.860	0.048	0.0	0.0	0.0	0.0	1.0-04	1.0
4244	8	154	160	2432.5	2.424	0.860	0.048	0.0	0.0	0.0	0.0	1.0-04	1.0
4245	8	154	161	2432.5	2.424	0.860	0.048	0.0	0.0	0.0	0.0	1.0-04	1.0
4246	8	158	159	0.0	2.758	3.275	0.0	0.0	0.0	0.0	0.0	3.542	2.55
4247	8	159	160	0.0	2.758	3.275	0.0	0.0	0.0	0.0	0.0	3.542	2.55
4248	8	160	161	0.0	2.758	3.275	0.0	0.0	0.0	0.0	0.0	3.542	2.55
4249	8	161	158	0.0	2.758	3.275	0.0	0.0	0.0	0.0	0.0	3.542	2.55
4250	8	153	95	2991.627	0.654	1.178	0.819	0.0	0.0	0.0	0.0	1.0-03	0.25
4251	8	153	101	2991.627	0.654	1.178	0.819	0.0	0.0	0.0	0.0	1.0-03	0.25
*****													
* OUTLET ANNULUS FLOW PATHS													
*****													
4252	8	154	162	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4253	8	154	163	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4254	8	154	164	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4255	8	154	165	974.383	0.368	2.640	3.46	0.0	0.0	0.0	0.0	1.0-03	1.833
4256	8	162	166	3383.84	2.252	2.005	1.672	0.0	0.0	0.0	0.0	1.0-04	1.333
4257	8	163	166	3383.84	2.252	2.005	1.672	0.0	0.0	0.0	0.0	1.0-04	1.333
4258	8	164	166	3383.84	2.252	2.005	1.672	0.0	0.0	0.0	0.0	1.0-04	1.333
4259	8	165	95	2559.52	1.5	3.010	6.586	0.0	0.0	0.0	0.0	1.0-04	1.333
4260	8	163	162	0.0	1.106	2.722	0.0	0.0	0.0	0.0	0.0	2.442	1.233
4261	8	163	164	0.0	1.106	2.722	0.0	0.0	0.0	0.0	0.0	2.442	1.233
4262	8	164	165	0.0	1.383	2.722	0.0	0.0	0.0	0.0	0.0	3.052	1.233
4263	8	165	168	0.0	1.659	2.722	0.0	0.0	0.0	0.0	0.0	3.663	1.233
4264	8	166	95	5075.76	2.741	3.295	0.0	0.0	0.0	0.0	0.0	1.0-03	1.248
4265	8	167	95	5075.76	2.741	3.295	0.0	0.0	0.0	0.0	0.0	1.0-03	1.248
4266	8	167	101	5075.76	2.741	3.295	0.0	0.0	0.0	0.0	0.0	1.0-03	1.248
4267	8	166	101	5075.76	2.741	3.295	0.0	0.0	0.0	0.0	0.0	1.0-03	1.248
4268	8	162	80	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4269	8	164	80	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4270	8	165	82	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4271	8	168	84	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4272	8	169	86	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4273	8	171	86	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4274	8	172	88	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4275	8	173	78	0.0	0.233	1.069	0.0	0.0	1.0+16	0.0	0.0	1.0-03	1.167
4276	8	154	168	974.383	0.368	2.640	3.46	0.0	0.0	0.0	0.0	1.0-03	1.833
4277	8	154	169	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4278	8	154	170	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4279	8	154	171	2327.08	0.501	6.305	3.49	0.0	0.0	0.0	0.0	1.0-03	2.833
4280	8	154	172	974.383	0.368	2.640	3.46	0.0	0.0	0.0	0.0	1.0-03	1.833
4281	8	154	173	974.383	0.368	2.640	3.46	0.0	0.0	0.0	0.0	1.0-03	1.833
4282	8	169	168	0.0	1.383	2.722	0.0	0.0	0.0	0.0	0.0	3.052	1.233
4283	8	170	169	0.0	1.106	2.722	0.0	0.0	0.0	0.0	0.0	2.442	1.233
4284	8	170	171	0.0	1.106	2.722	0.0	0.0	0.0	0.0	0.0	2.442	1.233
4285	8	171	172	0.0	1.383	2.722	0.0	0.0	0.0	0.0	0.0	3.052	1.233
4286	8	172	173	0.0	1.659	2.722	0.0	0.0	0.0	0.0	0.0	3.663	1.233
4287	8	162	173	0.0	1.383	2.722	0.0	0.0	0.0	0.0	0.0	3.052	1.233
4288	8	159	168	1584.93	1.743	0.876	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4289	8	160	171	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4290	8	160	170	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4291	8	160	169	1056.62	2.615	0.584	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4292	8	161	173	1584.93	1.743	0.876	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230
4293	8	161	172	1584.93	1.743	0.876	0.059	0.0	0.0	0.0	0.0	1.0-03	0.230

Table 2.26-2. (Cont'd)

****	4294	8,169,167,3383.84,2.252,2.005,1.672,0.0,0.0,0.0,1.0-04,1.333
****	4295	8,173,101,2559.52,1.5,3.010,6.536,0.0,0.0,0.0,1.0-04,1.333
****	4296	8,172,101,2559.52,1.5,3.010,6.536,0.0,0.0,0.0,1.0-04,1.333
****	4297	8,27,29,0.0,0.141,38.644,0.0,0.0,0.0,0.0,0.707,7.014
****	4298	8,23,30,0.0,0.141,33.644,0.0,0.0,0.0,0.0,0.707,7.014
****	4299	8,1,3,1.826+04,1.362,7.069,0.1826,0.0,0.0,0.0,5.043-03,3.0
****	4300	8,2,4,1.826+04,1.362,7.069,0.1826,0.0,0.0,0.0,5.043-03,3.0
****	4301	8,3,5,1.826+04,1.369,7.069,0.0,0.0,0.0,0.0,0.0276,3.0
****	4302	8,4,6,1.826+04,1.369,7.069,0.0,0.0,0.0,0.0,0.0276,3.0
		*****
		* LEAK FLOW PATHS
****	4303	7,63,174,-10,-10,4.276,1.21,1.0,0.0
****	4304	7,62,174,-10,-10,4.276,1.21,1.0,0.0
		*****
		* MODIFIED FLOW PATHS
****	4901	194,622.052,622.052,195,622.052,622.052,196,622.052,622.052
****	4902	197,622.052,622.052,198,622.052,622.052,199,622.052,622.052
****	4903	200,622.052,622.052,201,622.052,622.052,202,622.052,622.052
****	4904	203,622.052,622.052,204,622.052,622.052,205,622.052,622.052
****	4905	209,609.875,609.875,209,609.875,609.875,210,609.875,609.875
****	4906	211,609.875,609.875,212,609.875,609.875,213,609.875,609.875
****	4907	214,610.125,610.125,219,610.037,610.037,222,622.936,622.936
****	4908	226,614.527,614.527,227,617.277,617.277,228,620.443,620.443
****	4909	297,601.011,601.011,298,601.011,601.011,81,623.244,628.465
		*****
		* VARIABLE FLOW AREA VS. TIME TABLE
****	4951	4,270,0,4.062,0.062,3.635,0.0116,2.993,0.0159,2.138,0.020,1.069,0.0242
****		+0.0,0.0271,0.0,1.0+06
		*****
		* VENT VALVE SPECIFICATIONS
****	4820	268,269,270,271,272,273,274,275
****	4821	0,625,0,742,0,167,0,037,0,454,2,558,113,11,0,0,1,0
****	4831	1,0+06,0,1,0+06,0373,1,0+04,105,630,124,315,129,153,140,86
****		+4,157,62.1,175,46.8,192,29,227,21,262,16,297,11,332,9.7,36
****		+7,8.6,401,6.3,456,5.7,4537,5.7,1.57
		*****
		* CORE PARAMETERS
****	5001	0,048,2,0,7500,0,3000,0,10,0,1,175583.71,8296.4,0.166,-1.2-03
****	5002	1,0,1,0,0,0,0,0,0
****	5003	0,048,4,0,7500,0,3000,0,10,0,1,175583.71,16592.8,0.33,-1.2-03
****	5004	1,0,1,0,0,0,0,0
****	5005	0,048,4,0,7500,0,3000,0,10,0,1,175583.71,16592.8,0.33,-1.2-03
****	5006	1,0,1,0,0,0,0,0
****	5007	0,048,2,0,7500,0,3000,0,10,0,1,175583.71,8296.4,0.166,-1.2-03
****	5008	1,0,1,0,0,0,0,0
		*****
		* FUEL PTN GAP AND CLADDING PROPERTIES
****	5101	0,0152775,0,0,1325,0,0004657,-1,-1,0021894
****	5102	0,0,0,0,1,1,775
****	5103	0,0152775,0,0,1325,0,0004657,-1,-1,0021894
****	5104	0,0,0,0,1,1,775
****	5105	0,0152775,0,0,1325,0,0004657,-1,-1,0021894
****	5106	0,0,0,0,1,1,775
****	5107	0,0152775,0,0,1325,0,0004657,-1,-1,0021894
****	5108	0,0,0,0,1,1,775
****	5500	660,26
****	5505	0,35,0,00020307,0,00050972,1,0,2,0153,1
****	5510	1,1,4

Table 2.26-2. (Cont'd)

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*****
* FORCE ON CORE
*****
5911,4,40.99,49.64,267624.0
*****
* SCRAM PARAMETERS
*****
6001,0.0,1.0+04,2064.7,0.5,1.0+04,21,21
6011,1.0,0.0,1.0,100.
*****
* STEAM GENERATOR PARAMETERS
*****
7001,17,242781.875,525.0
7002,18,242781.875,525.0
7003,19,242781.875,525.0
7004,20,242781.875,525.0
7005,21,242781.875,525.0
7006,22,242781.875,525.0
7007,23,242781.875,525.0
7008,24,242781.875,525.0
7009,25,242781.875,525.0
7010,26,242781.875,525.0
7011,1.0-05,1.0+06,14.70,1.0+06,1.0+06,26,26
7021,1.0,0.0,1.0,1.0+06
*****
* PUMP CHARACTERISTICS
*****
8001,1,2.73,2.37,2.07,1.85,1.69,1.51,1.40,1.25,1.17,1.11,1.
8002,2,2.71,2.47,2.29,2.14,1.95,1.82,1.70,1.58,1.48,1.41,1.
8003,3,2.44,2.30,2.16,2.03,1.93,1.83,1.74,1.65,1.57,1.50,1.
8004,4,2.71,2.69,2.54,2.38,2.24,2.11,2.00,1.90,1.81,1.73,1.
8005,5,1.96,1.56,1.25,1.13,1.03,0.93,0.84,0.77,0.71,0.66,1.
8006,6,4.1,1.0,1.8,3.8,5.3,8.7,1.29,1.98,2.87,3.7,4.73
8007,7,4.73,4.31,3.91,3.51,3.13,2.78,2.43,2.11,1.81,1.53,1.
8008,8,4.1,0.60,0.78,0.91,1.02,1.13,1.21,1.34,1.50,1.66,1.96
*****
* PUMP ELECTRIC TORQUE VS. ACTUAL PUMP SPEED
*****
8016,40162.0,41344.120,42919.240,44100.360,46069.480,48038.
8017,600,50794.720,55125.840,62212.960,82688.1080,88988.1140.
8018,74025.1176.0,1200.-74025.1224.-88988.1260.-82688.1320.
8019,-62212.1440.-55125.1560.-50794.1680.-48038.1800.46069.
8020,1920.-44100.2040.-42919.2160.-41344.2280.-40162.2400.
*****
* PUMP SHUTDOWN PARAMETERS
*****
8021,0,1,+10,0,0,0,0,0,0,0,0,0
8022,0,1,+10,0,0,0,0,0,0,0,0,0
8023,0,1,+10,0,0,0,0,0,0,0,0,0
8024,0,1,+10,0,0,0,0,0,0,0,0,0
*****
* PUMP PARAMETERS
*****
8031,45,273,37300,1191,88068.45,4.28,1191.
8032,46,273,37300,1191,88068.45,4.28,1191.
8033,47,273,37300,1191,88068.45,4.28,1191.
8034,48,273,37300,1191,88068.45,4.28,1191.
*****
* PUMP MOTOR CONSTANTS
*****
8041,71,111,429.34,40000,117.34,1.7372,74.048,4869.390,115
8042,00102,4332,1925,71100.
8051,-26.6026,50.9643,-33.6919,9.3302

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INPUT FILE OCCUPIES 8809 DATA WORDS

Table 2.26-3. Listing of CRAFT2 Code Inputs for Nozzle-Supported Reactor Vessel Internals ΔP Model

LISTING OF INPUT DATA FOR CASE 1

* CASE INFORMATION	
* CALC. 32-110239-01	
* MODIFIED BREAK OPENING AREA VS. TIME SCHEDULE	
* TOLEDO NS14	
* .5 SEC KVA	
* BREAK OPENING TIME=.070 SEC.	
****	1001,0.50,7.0-05,0.070
****	1002,0.0,0.0035,0.025,0.042,0.009,0.0554,0.016,0.060,0.026,0.0813,0.05,0.0936,0.045,0.0968,0.053,0.121,0.070,0.121,1.0+06
****	1004,5.5,5.5
* PLOT INFORMATION	
****	1 1501,1101,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	2 + 1102,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	3 + 1103,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	4 + 1104,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	5 + 1105,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	6 + 1106,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	7 + 1107,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	8 + 1108,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	9 + 1109,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	10 + 1110,1.0,0.5,0.1,1.0,0.0,5.0,0,0
****	11 1502,1111,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	12 + 1112,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	13 + 1113,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	14 + 1114,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	15 + 1115,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	16 + 1116,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	17 + 1117,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	18 + 1118,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	19 + 1119,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	20 + 1120,1.0,0.5,0.1,1.0,0.0,5.0,0,0
****	21 1503,1121,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	22 + 1122,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	23 + 1123,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	24 + 1124,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	25 + 1125,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	26 + 1126,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	27 + 1127,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	28 + 1128,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	29 + 1129,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	30 + 1130,1.0,0.5,0.1,1.0,0.0,5.0,0,0
****	31 1504,1131,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	32 + 1132,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	33 + 1133,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	34 + 1134,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	35 + 1135,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	36 + 1136,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	37 + 1137,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	38 + 1138,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	39 + 1139,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	40 + 1140,1.0,0.5,0.1,1.0,0.0,5.0,0,0
****	41 1505,1141,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	42 + 1142,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	43 + 1143,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	44 + 1144,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	45 + 1145,1.0,0.5,0.1,1.0,0.0,5.0,0,0
	46 + 1146,1.0,0.5,0.1,1.0,0.0,5.0,0,0

Table 2.26-3. (Cont'd)

47	+	1147	1.0	0.5	0.1	1.0	0.5	0.0	0.0
48	+	1143	1.0	0.5	0.1	1.0	0.5	0.0	0.0
49	+	1149	1.0	0.5	0.1	1.0	0.5	0.0	0.0
50	+	1150	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1506	1151	1.0	0.5	0.1	1.0	0.5	0.0	0.0
52	+	1152	1.0	0.5	0.1	1.0	0.5	0.0	0.0
53	+	1153	1.0	0.5	0.1	1.0	0.5	0.0	0.0
54	+	1154	1.0	0.5	0.1	1.0	0.5	0.0	0.0
55	+	1155	1.0	0.5	0.1	1.0	0.5	0.0	0.0
56	+	1156	1.0	0.5	0.1	1.0	0.5	0.0	0.0
57	+	1157	1.0	0.5	0.1	1.0	0.5	0.0	0.0
58	+	1158	1.0	0.5	0.1	1.0	0.5	0.0	0.0
59	+	1159	1.0	0.5	0.1	1.0	0.5	0.0	0.0
60	+	1160	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1507	1161	1.0	0.5	0.1	1.0	0.5	0.0	0.0
62	+	1162	1.0	0.5	0.1	1.0	0.5	0.0	0.0
63	+	1163	1.0	0.5	0.1	1.0	0.5	0.0	0.0
64	+	1164	1.0	0.5	0.1	1.0	0.5	0.0	0.0
65	+	1165	1.0	0.5	0.1	1.0	0.5	0.0	0.0
66	+	1166	1.0	0.5	0.1	1.0	0.5	0.0	0.0
67	+	1167	1.0	0.5	0.1	1.0	0.5	0.0	0.0
68	+	1168	1.0	0.5	0.1	1.0	0.5	0.0	0.0
69	+	1169	1.0	0.5	0.1	1.0	0.5	0.0	0.0
70	+	1170	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1508	1171	1.0	0.5	0.1	1.0	0.5	0.0	0.0
72	+	1172	1.0	0.5	0.1	1.0	0.5	0.0	0.0
73	+	1173	1.0	0.5	0.1	1.0	0.5	0.0	0.0
74	+	1174	1.0	0.5	0.1	1.0	0.5	0.0	0.0
75	+	1175	1.0	0.5	0.1	1.0	0.5	0.0	0.0
76	+	1176	1.0	0.5	0.1	1.0	0.5	0.0	0.0
77	+	1177	1.0	0.5	0.1	1.0	0.5	0.0	0.0
78	+	1178	1.0	0.5	0.1	1.0	0.5	0.0	0.0
79	+	1179	1.0	0.5	0.1	1.0	0.5	0.0	0.0
80	+	1180	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1509	1181	1.0	0.5	0.1	1.0	0.5	0.0	0.0
82	+	1182	1.0	0.5	0.1	1.0	0.5	0.0	0.0
83	+	1183	1.0	0.5	0.1	1.0	0.5	0.0	0.0
84	+	1184	1.0	0.5	0.1	1.0	0.5	0.0	0.0
85	+	1185	1.0	0.5	0.1	1.0	0.5	0.0	0.0
86	+	1186	1.0	0.5	0.1	1.0	0.5	0.0	0.0
87	+	1187	1.0	0.5	0.1	1.0	0.5	0.0	0.0
88	+	1188	1.0	0.5	0.1	1.0	0.5	0.0	0.0
89	+	1189	1.0	0.5	0.1	1.0	0.5	0.0	0.0
90	+	1190	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1510	1191	1.0	0.5	0.1	1.0	0.5	0.0	0.0
92	+	1192	1.0	0.5	0.1	1.0	0.5	0.0	0.0
93	+	1193	1.0	0.5	0.1	1.0	0.5	0.0	0.0
www	1511	1801	1.0	0.5	0.1	1.0	0.5	0.0	5.10
95	+	1802	1.0	0.5	0.1	1.0	0.5	0.0	10.84
96	+	1803	1.0	0.5	0.1	1.0	0.5	0.0	10.69
97	+	1804	1.0	0.5	0.1	1.0	0.5	0.0	10.75
98	:	1805	1.0	0.5	0.1	1.0	0.5	0.0	6.38
99	+	1806	1.0	0.5	0.1	1.0	0.5	0.0	10.85
100	+	1807	1.0	0.5	0.1	1.0	0.5	0.0	10.70
101	+	1808	1.0	0.5	0.1	1.0	0.5	0.0	10.67
102	+	1809	1.0	0.5	0.1	1.0	0.5	0.0	39.39
103	+	1810	1.0	0.5	0.1	1.0	0.5	0.0	10.86
104	+	1811	1.0	0.5	0.1	1.0	0.5	0.0	10.71

Table 2.26-3. (Cont'd)

105	+	1812	1.0	0.5	0.1	1.0	0.5	0.5	5.9					
106	+	1813	1.0	0.5	0.1	1.0	0.5	10.87						
107	+	1814	1.0	0.5	0.1	1.0	0.5	10.72						
108	+	1815	1.0	0.5	0.1	1.0	0.5	9.6						
****	109	1512	2520	1.0	0.5	0.1	1.0	0.5	0.0					
110	+	2516	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
111	+	2561	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
112	+	2579	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
113	+	2585	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
114	+	2187	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
115	+	2188	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
116	+	2190	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
117	+	2186	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
118	+	2185	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
119	+	2125	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
120	+	2128	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
****	121	1513	21135	1.0	0.5	0.1	1.0	0.5	0.0					
122	+	21139	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
123	+	21140	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
124	+	21141	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
125	+	21131	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
126	+	21103	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
127	+	21104	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
128	+	21105	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
129	+	21106	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
****	130	1514	5701	1.0	0.5	0.1	1.0	0.5	0.0					
131	+	5401	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
132	+	5501	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
133	+	6259	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
134	+	6278	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
135	+	2159	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
136	+	2178	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
137	+	8559	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
138	+	8578	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
139	+	2259	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
140	+	2278	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
141	+	1194	1.0	0.5	0.1	1.0	0.5	0.0	0.0					
****	1601	1105	1103	1108	1106	1136	1110	1173	1188	1801	1805	5701	1809	1812
****	1701	1	1											
****		* TIME STEPS												
****	2001	00007	2	100	00007	1	80							
****		* CONTROL VOLUMES												
****		*CARD	AREA	HEIGHT	EXIT	INLET	BOTTOM	PRESSURE	ENTHALPY	LEVEL				
****	3001	8.85	0.4	705	563.42	567.42	567.92	2225.278	553.68	0.				
****	3002	8.8550	4.705		533.42	567.42	567.92	2225.278	553.68	0.				
****	3003	8.5560	4.75		553.96	557.71	553.46	2228.458	553.68	0.				
****	3004	8.5560	4.75		553.96	557.71	553.46	2228.458	553.68	0.				
****	3005	51.684	7.1670		546.79	552.96	546.29	2227.430	553.68	0.				
****	3006	73.479	1.4790		552.83	552.35	551.85	2221.330	553.68	0.				
****	3007	49.925	3.75		557.71	554.66	554.46	2213.383	559.4	0.0				
****	3008	49.925	4.71		562.42	558.71	558.21	2208.742	590.3	0.0				
****	3009	55.197	2.4370		551.35	549.92	549.42	2221.124	553.68	0.				
****	3010	82.092	6.00		578.46	578.46	577.96	2190.500	625.74	0.				
****	3011	4.1230	3.00		571.0	571.0	569.5	2163.428	625.74	0.				
****	3012	4.1230	3.00		571.0	571.0	569.5	2163.428	625.74	0.				
****	3013	6.3390	5.50		571.0	571.0	568.0	2177.000	625.74	0.				
****	3014	6.3390	5.50		571.0	571.0	568.0	2177.000	625.74	0.				
****	3015	15.116	8.0		638.48	639.98	637.98	2133.329	625.74	0.				

Table 2.26-3. (Cont'd)

3016	15.116	8.0	638.48	639.98	637.98	2133.329	625.74	0.
3017	7.0480	31.74	607.24	576.5	576.0	2173.831	625.74	0.
3018	7.0480	31.74	607.24	576.5	576.0	2173.831	625.74	0.
3019	15.330	6.5	575.5	571.0	569.5	2162.009	625.74	0.
3020	15.330	6.5	575.5	571.0	569.5	2162.009	625.74	0.
3021	26.155	11.24	575.0	535.0	574.452	2137.50	553.68	0.0
3022	26.155	11.24	620.0	630.0	619.412	2144.04	624.573	0.0
3023	26.155	11.24	575.0	535.0	574.452	2137.50	553.68	0.0
3024	26.155	11.24	620.0	630.0	619.412	2144.04	624.573	0.0
3025	10.+5	10.+2	571.0	571.0	0.0	14.7	0.0	1.-5
3026	3.34	3.033	571.0	571.0	569.853	2227.125	553.680	0.0
CHIENUPD3027	5.4242	3.0333	571.0	571.0	569.853	2227.161	553.68	0.
CHIENUPD3028	5.4242	3.0333	571.0	571.0	569.853	2227.161	553.68	0.
CHIENUPD3029	5.4242	3.0333	571.0	571.0	569.853	2227.161	553.68	0.
3030	7.0480	31.74	607.24	607.74	607.74	2137.973	625.74	0.
3031	2.3560	3.00	571.0	571.0	569.5	2163.363	625.74	0.
3032	12.894	4.75	557.71	553.94	553.46	2215.500	553.68	0.
3033	14.103	4.71	562.42	553.71	553.71	2215.500	590.30	0.
3034	11.398	6.92	574.0	568.03	567.50	2134.350	553.24	0.
3035	11.398	6.92	574.0	568.03	567.50	2134.350	553.24	0.
3036	23.276	5.84	570.33	575.17	569.83	2224.830	553.68	0.
3037	22.18	5.84	570.33	575.2	569.830	2224.830	553.680	0.0
3038	55.653	1.125	553.96	553.83	553.33	2213.000	553.68	0.
3039	64.106	2.292	568.25	566.96	566.45	2213.000	625.48	0.
3040	11.398	6.92	574.0	568.03	567.50	2134.350	553.24	0.
3041	11.398	6.92	574.0	568.03	567.50	2134.350	553.24	0.
3042	23.276	5.84	570.33	575.17	569.83	2224.830	553.68	0.
3043	23.276	5.84	570.33	575.17	569.83	2224.830	553.68	0.
3044	77.526	56.2	601.57	601.57	573.47	1000.	0.0	25.
3045	77.526	56.2	601.57	601.57	573.47	1000.	0.0	25.
3046	24.100	44.0	593.0	593.0	592.0	2148.440	0.000	8.0
3047	53.989	4.92	569.05	572.97	568.55	2141.557	553.24	0.
3048	53.989	4.92	569.05	572.97	568.55	2141.557	553.24	0.
3049	37.645	7.33	631.17	637.48	630.65	2145.146	625.74	0.
3050	37.645	7.33	631.17	637.48	630.65	2145.146	625.74	0.
3051	7.0600	31.74	638.98	603.24	607.74	2137.973	625.74	0.
3052	8.8550	4.705	563.42	567.42	562.92	2225.778	553.68	0.
3053	8.8550	4.75	553.94	557.71	553.46	2229.458	553.68	0.
3054	8.8550	4.705	563.42	567.42	562.92	2225.278	553.68	0.
3055	8.8550	4.75	553.96	557.71	553.46	2228.458	553.68	0.
3056	49.925	3.54	565.96	563.42	562.92	2205.000	620.13	0.0
3057	13.422	4.71	567.12	563.42	562.92	2205.400	620.13	0.
3058	3.1660	4.75	557.71	553.96	553.46	2232.446	553.68	0.
3059	2.9920	4.71	562.42	558.71	558.21	2231.925	553.68	0.
3060	3.4350	4.71	567.12	563.42	562.92	2230.404	553.68	0.
3061	8.6290	4.71	558.71	562.42	558.21	2226.867	553.68	0.
3062	8.6290	4.71	558.71	562.42	558.21	2226.867	553.68	0.
3063	8.6290	4.71	558.71	562.42	558.21	2226.867	553.68	0.
3064	8.6290	4.71	558.71	562.42	558.21	2226.867	553.68	0.
CHIENUPD3065	55.8355	4.75	571.0	571.0	568.75	2193.220	625.74	0.
3066	23.348	4.6040	572.85	569.25	568.75	2194.425	625.74	0.
3067	23.348	4.6040	577.46	573.85	573.35	2193.223	625.74	0.
3068	2.3560	3.00	571.0	571.0	569.5	2163.363	625.74	0.
CHIENUPD3069	4.486	4.458	575.72	575.72	573.5	2190.750	625.74	0.
CHIENUPD3070	4.486	4.458	575.729	575.729	573.5	2190.750	625.74	0.
CHIENUPD3071	4.486	4.458	575.729	575.729	573.5	2190.750	625.74	0.
CHIENUPD3072	4.486	4.458	575.729	575.729	573.5	2190.750	625.74	0.
3073	2.75	2.330	571.0	571.0	569.830	2224.490	553.680	0.0

Table 2.26-3. (Cont'd)

3074	26.155	11.24	609.0	619.0	608.172	2142.94	621.741	0.0												
3075	55.8355	4.4513	575.7292	575.7292	573.5	2120.521	625.74	0.0												
CHIENUPD3076	4.3110	5.500	570.75	570.75	570.0	2120.521	625.74	0.0												
CHIENUPD3077	4.3110	5.500	570.75	570.75	570.0	2120.521	625.74	0.0												
CHIENUPD3078	4.3110	5.500	570.75	570.75	570.0	2120.521	625.74	0.0												
CHIENUPD3079	4.3110	5.500	570.75	570.75	570.0	2120.521	625.74	0.0												
CHIENUPD3030	6.4332	5.875	571.0	571.0	567.625	2229.702	553.68	0.0												
CHIENUPD3031	6.4332	5.875	571.0	571.0	567.625	2229.702	553.68	0.0												
CHIENUPD3032	6.4332	5.875	571.0	571.0	567.625	2229.702	553.68	0.0												
CHIENUPD3033	6.4332	5.875	571.0	571.0	567.625	2229.702	553.68	0.0												
CHIENUPD3034	6.4332	3.333	575.167	575.167	573.5	2229.655	553.68	0.0												
CHIENUPD3035	6.4332	3.333	575.167	575.167	573.5	2229.655	553.68	0.0												
CHIENUPD3036	6.4332	3.333	575.167	575.167	573.5	2229.655	553.68	0.0												
CHIENUPD3037	6.4332	3.333	575.167	575.167	573.5	2229.655	553.68	0.0												
3080	2.75	2.330	571.0	571.0	569.030	2224.500	553.600	0.0												
3089	26.155	11.24	597.0	608.0	596.932	2141.88	594.012	0.0												
3090	26.155	11.24	596.0	596.0	595.6	2141.88	597.407	0.0												
3091	26.155	11.24	609.0	619.0	608.172	2142.94	621.741	0.0												
3092	26.155	11.24	597.0	608.0	596.932	2141.88	594.012	0.0												
3093	26.155	11.24	596.0	596.0	595.692	2141.89	597.407	0.0												
3001	25.5978																			
*BUBBLE RISE VELOCITY																				
3901	1028.0																			
CARD, TY, FR, TO, FLOW, L/A, AREA, FWDR, MOMFFAC, REWK, HY, KFRIC, HYDIA																				
4001	1, 38	7, 35827.80	0.0475	49.90	2.529	0.0	0.0	0.0	0.0	0.0	0.695	0.0478								
4002	1, 7	8, 35827.80	0.0440	49.90	2.372	-3.784	9.0	0.0	0.0	0.0	1.234	0.0478								
4003	1, 8	56, 35827.80	0.0340	49.90	2.153	-4.233	9.0	0.0	0.0	0.0	0.761	0.0478								
4004	1, 56	39, 35827.80	0.0526	49.90	1.708	0.0	0.0	0.0	0.0	0.0	1.144	0.0478								
CHIENUPD4005	03, 81	52, 9033.32	0.7776	7.374	2.054	4.97	-7.0	0.0	0.0	1	-5	1.42								
CHIENUPD4006	03, 82	54, 9033.32	0.7776	7.374	2.054	4.97	-7.0	0.0	0.0	1	-5	1.42								
CHIENUPD4007	03, 80	01, 9033.32	0.7776	7.374	2.054	4.97	-7.0	0.0	0.0	1	-5	1.42								
CHIENUPD4008	03, 83	02, 9033.32	0.7776	7.374	2.054	4.97	-7.0	0.0	0.0	1	-5	1.42								
4009	08, 61	64, 0.0	2.73	3.85	1	-10	0.0	0.0	0.0	0.0	7.57	1.0								
4010	08, 06	33, 35827.5	0.0220	101.6	7.09	-3.01	-8.0	0.0	0.0	0.0	1	-10	1.0							
4011	08, 74	72, 18238.9	0.430	24.2	0.134	0.0	0.0	0.0	0.0	0.0	3.43	1.0								
4012	08, 61	03, 9033.3	0.547	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4013	08, 01	61, 9033.3	0.542	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4014	08, 02	64, 9033.3	0.542	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4015	08, 12	31, 18238.9	0.191	7.07	1	-10	0.0	0.0	0.0	0.0	0.0382	1.0								
4016	08, 32	33, 650.0	0.240	14.8	1463.5	0.0	0.0	0.0	0.0	1.0	-4	1.0								
4017	08, 62	53, 9033.3	0.547	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4018	08, 63	55, 9033.3	0.547	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4019	08, 64	04, 9033.3	0.547	8.61	1	-10	0.0	0.0	0.0	0.0	0.027	1.0								
4020	08, 04	32, 650.0	0.240	14.8	1	-10	3.076	-6.0	0.0	0.0	8.59	1.0								
4021	08, 03	05, 9033.3	1.66	8.61	1.83	1.22	-6.0	0.0	0.0	0.0	1	-10	1.0							
4022	08, 04	05, 9033.3	1.66	8.61	1.83	1.22	-6.0	0.0	0.0	0.0	1	-10	1.0							
4023	08, 05	09, 36333.2	0.0169	53.5	362	-3.44	-8.0	0.0	0.0	0.0	1	-10	1.0							
4024	08, 61	62, 0.0	2.73	3.85	1	-10	0.0	0.0	0.0	0.0	7.57	1.0								
4025	08, 13	11, 18238.9	0.276	9.0841	-1.55	-6.0	0.0	0.0	0.0	0.0	0.0391	1.0								
4026	08, 14	12, 18238.9	0.276	9.0841	-1.55	-6.0	0.0	0.0	0.0	0.0	0.0391	1.0								
4027	08, 17	30, 18238.9	4.49	7.07	1	-10	0.0	0.0	0.0	0.0	0.0905	1.0								
CARD, TY, FR, TO, FLOW, L/A, AREA, FWDR, MOMFFAC, REWK, HT, KFRIC, HYDIA																				
4028	08, 18	51, 18238.9	4.49	7.07	1	-10	0.0	0.0	0.0	0.0	0.0905	1.0								
CHIENUPD4029	03, 82	60, 36120.3	4.125	0.7963	471.97	+0.039	10.0	0.0	0.0	0.0	1	-5	0.167							
4030	08, 31	19, 18238.9	0.954	7.07	1	-10	0.0	0.0	0.0	0.0	0.0192	1.0								
4031	08, 11	68, 18238.9	0.191	7.07	1	-10	0.0	0.0	0.0	0.0	0.0382	1.0								
4032	08, 19	17, 18238.9	3.356	7.07	0.182	0.0	0.0	0.0	0.0	0.0	0.045	1.0								
4033	08, 20	18, 18238.9	3.356	7.07	0.182	0.0	0.0	0.0	0.0	0.0	0.045	1.0								



Table 2.26-3. (Cont'd)

4034	08	30	15	18238.9	4.453	7.07	1160	0.0	0.0	0.0	0.045	1.		
4035	08	51	16	18238.9	4.453	7.07	1160	0.0	0.0	0.0	0.045	1.		
4036	03	22	74	18238.9	0.430	26.2	0.136	0.0	0.0	3.43	1.			
4037	03	24	91	18238.9	0.430	26.2	0.133	0.0	0.0	3.43	1.			
4038	03	21	68	18238.9	0.271	24.2	0.977	0.0	0.0	1.66	1.			
4039	03	23	47	18238.9	0.238	26.2	0.977	0.0	0.0	1.66	1.			
4040	08	48	41	9119.44	1.593	4.28	143	-2.90	-6	0.0	0.0166	1.		
4041	03	47	40	9119.44	1.593	4.28	143	-2.90	-6	0.0	0.0166	1.		
4042	03	43	35	9119.44	1.593	4.28	143	-2.90	-6	0.0	0.0166	1.		
4043	08	47	34	9119.44	1.593	4.28	143	-2.90	-6	0.0	0.0166	1.		
4044	02	35	37	9119.44	3.257	4.28	9050	0.0	0.0	0.0	0.0265	1.		
4045	02	34	36	9119.44	3.257	4.28	9050	0.0	0.0	0.0	0.0265	1.		
4046	02	41	43	9119.44	3.257	4.28	9050	0.0	0.0	0.0	0.0265	1.		
4047	02	40	42	9119.44	3.257	4.28	9050	0.0	0.0	0.0	0.0265	1.		
4048	03	43	23	9119.44	2.69	4.28	178	1.98	-6	0.0	0.0328	1.		
4049	08	42	29	9119.44	2.69	4.28	178	1.98	-6	0.0	0.0328	1.		
4050	8	57	63	9119.44	1.79	4.276	112	0.0	0.0	0.221	2.33	1.		
4051	03	55	27	9119.44	2.69	4.28	178	1.98	-6	0.0	0.0328	1.		
4052	03	49	24	18238.9	0.238	26.2	0.977	0.0	0.0	1.66	1.			
4053	03	56	57	0.0	0.031	1.65	1	-10	0.0	0.0	0.543	0.115		
CHIENUPD	4054	03	82	83	0.0	5.817	4.16	301	0.0	0.0	5.0	7.37	1.42	
4055	03	52	54	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
CHIENUPD	4056	03	81	82	0.0	2.508	4.16	045	0.0	0.0	5.0	7.37	1.42	
CHIENUPD	4057	03	83	80	0.0	2.508	4.16	045	0.0	0.0	5.0	7.37	1.42	
4058	08	16	49	18238.9	1.709	7.07	226	1.70	-6	0.0	0.0043	1.		
4059	7	73	25	-10	-10	4.276	1.21	-1	0.0	0.0	0.0	0.0	0.0	
4060	03	07	03	36333.2	0.0281	40.6	177	3.56	-8	0.0	0.0	1	-10	1.
4061	03	33	57	650.0	0.24	14.8	665.97	0.0	0.0	0.0	1.0	-4	1.	
4062	03	53	55	0.0	2.71	3.88	1	-10	0.0	0.0	0.0	7.54	1.	
CHIENUPD	4063	03	80	81	0.0	5.817	4.16	301	0.0	0.0	5.0	7.37	1.42	
4064	08	62	63	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
4065	03	15	50	18238.9	1.709	7.07	0.226	1.70	-6	0.0	0.0	0.043	1.	
4066	03	64	63	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
4067	03	50	22	18238.9	0.238	26.2	0.977	0.0	0.0	1.66	1.			
4068	03	59	06	144.5	1.112	9.69	1927	0.0	0.0	0.0	0.0	0.0	0.0	1.
4069	08	43	17	0.0	1.29	5	417	1	-10	0.0	0.0	177.6	73	
4070	08	53	05	9083.3	1.66	8.61	1.83	1.22	-6	0.0	0.0	1	-10	1.
4071	08	55	05	9083.3	1.66	8.61	1.83	1.22	-6	0.0	0.0	1	-10	1.
4072	08	52	62	9083.3	0.542	8.61	1	-10	0.0	0.0	0.0	0.27	1.	
4073	08	54	63	9083.3	0.542	8.61	1	-10	0.0	0.0	0.0	0.27	1.	
4074	08	01	52	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
4075	08	03	53	0.0	2.71	3.88	1	-10	0.0	0.0	0.0	7.54	1.	
4076	08	02	54	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
4077	08	04	55	0.0	2.71	3.88	1	-10	0.0	0.0	0.0	7.54	1.	
4078	7	88	25	-10	-10	4.276	1.21	-1	0.0	0.0	0.0	0.0	0.0	1.
4079	08	57	39	650.0	0.24	14.8	890.0	5.98	-7	0.0	0.0	1	-10	1.
4080	08	08	33	0.0	0.23	2.67	1	-10	0.0	0.0	0.0	6767	092	
4081	08	01	02	0.0	2.73	3.85	1	-10	0.0	0.0	0.0	7.57	1.	
CARD	TY	FR	TO	FLOW	L/A	AREA	FNDK	MON	FFAC	REWK	HT	KFRIC	HYDI	
4082	08	03	04	0.0	2.71	3.88	1	-10	0.0	0.0	0.0	7.57	1.	
4083	08	59	58	144.5	1.112	9.69	1927	0.0	0.0	0.0	0.0	3.53	1.	
4084	08	60	59	144.5	1.0988	9.69	1	-10	0.0	0.0	0.0	3.5371	1.	
4085	8	39	65	26867.00	0.0618	49.90	0.0	-1.973	-7	0.0	0.0	0.0	0.0	1.19
4086	8	59	66	9610.78	0.0099	1.0	0.015	0.0	0.0	0.0	0.0	1	-10	1.0
4087	5	65	14	3285.42	1.41	1.18	0.816	0.0	0.0	0.0	0.25	1	0-5	0.250
4088	5	65	13	3285.42	1.41	1.18	0.816	0.0	0.0	0.0	0.25	1	0-5	0.250
4089	8	66	67	3906.86	0.197	1.0	0.009	0.0	0.0	0.0	0.0	1	-10	1.0
4090	8	67	75	1168.51	0.136	13.42	0.0	0.0	0.0	0.0	0.25	0.0	0.250	

Table 2.26-3. (Cont'd)

4091	8,10,69	1352.51	2.504	0.2727	0.015	0.0	0.0	0.0	0.0	1.0-5	0.167
4092	8,67,10	2739.35	0.050	1.0	0.054	0.0	0.0	0.0	0.0	1.0-5	0.167
4093	03,03,20	18238.9	0.9540	7.07	1.10	0.0	0.0	0.0	0.0	0.0192	1.
CHIENUPD4094	05,72,70	0.0	3.13	2.88	.045	0.0	0.0	0.0	0.0	4.0,6.98	1.29
4095	03,91,92	18238.9	0.430	26.2	0.135	0.0	0.0	0.0	0.0	3.43	1.
4096	03,89,90	18238.9	0.430	26.2	0.135	0.0	0.0	0.0	0.0	3.43	1.
4097	08,92,93	18238.9	0.430	26.2	0.135	0.0	0.0	0.0	0.0	3.43	1.
4098	08,90,21	18238.9	0.430	26.2	0.135	0.0	0.0	0.0	0.0	3.43	1.
4099	8,10,70	1352.51	2.504	0.2727	0.015	0.0	0.0	0.0	0.0	1.0-5	0.167
4100	8,10,71	1352.51	2.504	0.2727	0.015	0.0	0.0	0.0	0.0	1.0-5	0.167
4101	8,10,72	1352.51	2.504	0.2727	0.015	0.0	0.0	0.0	0.0	1.0-5	0.167
CHIENUPD4102	05,71,69	0.0	3.13	2.88	.045	0.0	0.0	0.0	0.0	4.0,6.98	1.29
4103	8,79,14	7476.74	1.146	9.82	8.947	0.0	0.0	0.0	0.0	1.0-5	1.290
4104	8,78,14	7476.74	1.146	9.82	8.947	0.0	0.0	0.0	0.0	1.0-5	1.290
4105	8,77,13	7476.74	1.146	9.82	8.947	0.0	0.0	0.0	0.0	1.0-5	1.290
4106	8,76,13	7476.74	1.146	9.82	8.947	0.0	0.0	0.0	0.0	1.0-5	1.290
CARD, TY, FR, TO, FLOW				AREA, FREQ, MON, FFAC, REV, HT						K, FRIC, HYDIA	
4107	8,75,28	9119.44	0.351	4.276	0.142	1.25	-6	0.0	2.0	0.05	2.33
CHIENUPD4108	08,81,60	36,120	3.4125	0.7963	471.97	+0.039	10	0.0	0.0	1.0-5	0.167
CHIENUPD4109	03,81,60	36,120	3.4125	0.7963	471.97	+0.039	10	0.0	0.0	1.0-5	0.167
CHIENUPD4110	03,81,60	36,120	3.4125	0.7963	471.97	+0.039	10	0.0	0.0	1.0-5	0.167
DUMYPATH4111	03,25,25	0.0	0.158	1.00	149.2	0.0	0.0	0.0	0.0	1.0-5	1.
DUMYPATH4112	03,25,25	0.0	0.158	1.00	149.2	0.0	0.0	0.0	0.0	1.0-5	1.
4113	8,65,75	26000.08	0.054	55.84	0.193	0.0	0.0	0.0	0.0	1.0-5	1.19
4114	8,71,76	7476.74	0.855	5.823	0.0	0.0	0.0	0.0	0.0	0.0	1.290
4115	8,72,77	7476.74	0.855	5.823	0.0	0.0	0.0	0.0	0.0	0.0	1.290
4116	8,70,78	7476.74	0.855	5.823	0.0	0.0	0.0	0.0	0.0	0.0	1.290
4117	8,69,79	7476.74	0.855	5.823	0.0	0.0	0.0	0.0	0.0	0.0	1.290
CHIENUPD4118	03,29,80	9119.44	0.112	4.276	0.0448	+1.98	16	-6	0.09	2.0	0.0124, 2.33
CHIENUPD4119	03,27,81	9119.44	0.112	4.276	0.0448	+1.98	16	-6	0.09	2.0	0.0124, 2.33
4120	8,26,82	9119.44	0.16	4.276	0.090	2.71	-6	0.09	2.0	0.01	2.33
CHIENUPD4121	08,28,83	9119.44	0.112	4.276	0.0448	+1.98	16	-6	0.09	2.0	0.0124, 2.33
CHIENUPD4122	03,80,84	0.0	0.715	6.433	0.0	0.0	0.0	0.0	0.0	3.25	0.0
CHIENUPD4123	03,81,85	0.0	0.715	6.433	0.0	0.0	0.0	0.0	0.0	3.25	0.0
CHIENUPD4124	03,82,86	0.0	0.715	6.433	0.0	0.0	0.0	0.0	0.0	3.25	0.0
CHIENUPD4125	08,83,87	0.0	0.715	6.433	0.0	0.0	0.0	0.0	0.0	3.25	0.0
CHIENUPD4126	08,84,85	0.0	4.42	2.36	.045	0.0	0.0	0.0	0.0	3.0,7.37	1.42
CHIENUPD4127	08,85,86	0.0	4.42	2.36	.045	0.0	0.0	0.0	0.0	3.0,7.37	1.42
CHIENUPD4128	03,86,87	0.0	4.42	2.36	.045	0.0	0.0	0.0	0.0	3.0,7.37	1.42
CHIENUPD4129	08,87,84	0.0	4.42	2.36	.045	0.0	0.0	0.0	0.0	3.0,7.37	1.42
4130	8,75,10	2671.69	0.094	4.78	0.0	0.0	0.0	0.0	0.0	0.0	0.167
4131	8,66,65	5703.92	0.0412	70.26	35.37	0.0	0.0	0.0	0.0	.25, 1.0-5	0.250
CHIENUPD4132	05,76,77	0.0	2.54	3.55	.045	0.0	0.0	0.0	0.0	5.0,6.98	1.29
CHIENUPD4133	05,77,78	0.0	2.54	3.55	.045	0.0	0.0	0.0	0.0	5.0,6.98	1.29
CHIENUPD4134	05,78,79	0.0	2.54	3.55	.045	0.0	0.0	0.0	0.0	5.0,6.98	1.29
CHIENUPD4135	05,79,76	0.0	2.54	3.55	.045	0.0	0.0	0.0	0.0	5.0,6.98	1.29
CHIENUPD4136	05,71,72	0.0	3.13	2.88	.045	0.0	0.0	0.0	0.0	4.0,6.98	1.29
CHIENUPD4137	05,69,70	0.0	3.13	2.88	.045	0.0	0.0	0.0	0.0	4.0,6.98	1.29
4138	8,75,69	6124.23	0.63	12.097	2.933	0.0	0.0	0.0	0.0	1.8, 1.0-5	1.833
4139	8,75,70	6124.23	0.63	12.097	2.933	0.0	0.0	0.0	0.0	1.8, 1.0-5	1.833
4140	8,75,72	6124.23	0.63	12.097	2.933	0.0	0.0	0.0	0.0	1.8, 1.0-5	1.833
4141	8,75,71	6124.23	0.63	12.097	2.933	0.0	0.0	0.0	0.0	1.8, 1.0-5	1.833
CHIENUPD4142	08,69,87	0.0	0.478	1.069	3.9	0.0	1	+14	1.17	1.0-5	1.17
CHIENUPD4143	08,71,84	0.0	0.478	1.069	3.9	0.0	1	+14	1.17	1.0-5	1.17
CHIENUPD4144	08,72,85	0.0	0.478	1.069	3.9	0.0	1	+14	1.17	1.0-5	1.17
CHIENUPD4145	08,70,86	0.0	0.478	1.069	3.9	0.0	1	+14	1.17	1.0-5	1.17
4146	08,93,23	18238.9	0.430	26.2	0.135	0.0	0.0	0.0	0.0	3.43	1.
4147	10,88,73	9119.44	0.351	4.276	1.0	-5	0.0	0.0	2.0	0.006	2.33, 1.

Table 2.26-3. (Cont'd)

****	4148	8.5358	0.0	0.1	1.0	1.0+10	0.0	0.0	0.1	-5	1.0	
****	4149	8.0453	0.0	0.1	1.0	1.0+10	0.0	0.0	0.1	-5	1.0	
****	4150	7.0353	0.0	0.1	1.0	1.0+10	0.0	0.0	0.1	-5	1.0	
****	4151	8.5353	0.0	0.1	1.0	1.0+10	0.0	0.0	0.1	-5	1.0	
****	4671	59.78										
****	4672	59.0	1.									
****	4673	78.0	1.									
****	CHIENUPD4320	142.143	144.145									
****	CHIENUPD4321	0.5333	0.617	0.156	0.087	0.454	1.3524	80.76	0.0	1.0		
****	9M 10F3	4031	1.46	0.1	0.4	0.0373	1.4	105.630	124.315	129.153	140.86.4.15	
****	9M 20F3	47.62	1.175	46.8	192.29	227.21	282.16	297.11	332.9.7	367.8.6		
****	9M 30F3	4.01	0.3	433.5.7	4537.5.7	1.57						
****	4951	4.276	0.4	248.0025	4.096	009	4.039	016	4.019	026	3.928.033.3.	
****	+876	0.045	3.082	053.3.759	070.3.759	1.0+06						
****	* REACTION CORE PARAMETERS											
****	5001	0.4364	1.875	7500.	3000.	10.	3	191556.2	7771.0	0.3995	-1.157-3.5241	
****	5002	1.5241	1.0									
****	5003	0.4364	4.208	7500.	3000.	10.	3	191556.2	17441.	4727	-1.216-3.1.210	
****	5004	1.1.210	1.0									
****	5005	0.4364	4.167	7500.	3000.	10.	3	191556.2	17269.	4515	-1.216-3.1.189	
****	5006	1.1.189	1.0									
****	5007	0.4364	1.667	7500.	3000.	10.	3	191556.2	6908.0	0.363	-1.157-3.5328	
****	5008	1.5328	1.0									
****	* FUEL PIN GAP AND CLADDING PROPERTIES											
****	5101	0.1528	0.0	0.970	0.4	5.175-4	-1.	-1.	0.022053			
****	5102	0.0	0.0	1.1107								
****	5103	0.1533	0.0	0.1570	4.0175-4	-1.	-1.	0.022053				
****	5104	0.0	0.0	1.1107								
****	5105	0.1533	0.0	0.1570	4.0175-4	-1.	-1.	0.022053				
****	5106	0.0	0.0	1.1107								
****	5107	0.1528	0.0	0.970	0.4	5.175-4	-1.	-1.	0.022053			
****	5108	0.0	0.0	1.1107								
****	* FUEL PIN FUEL DENSITY											
****	5500	660.282										
****	5505	35.1.5527-4	2.8906-4	1.0	0.2	0.153	1					
****	5510	1.1.4										
****	* METAL WATER REACTION CONSTANTS											
****	5901	91.22	140.5	6.5	2.0	1.2500-6	1500.0	45500.0	33.3	0.0		
****	5911	4.41	70.43	93.27	4350.							
****	* DNB PARAMETERS											
****	5921	6.0+4	0.1	0.45	2.75	5						
****	* SCRAM AND KINETICS											
****	* SCRAM PARAMETERS											
****	6001	0.0	1.0+4	1900.0	0.5	1.0+4	21.21					
****	* KINETICS PARAMETERS											
****	6002	.283-4	.9675	.002	10000000.							
****	* MODERATOR COEFFICIENT=0.0 BOL											
****	6003	-1.45	0.-0.273	22.5	-0.174	30.-0.087	40.-0.053	50.				
****	6004	-0.032	60.-0.0195	67.-0.0156	70.-0.0102	75.						
****	6005	-0.0003	80.-0.0035	85.-0.0027	90.-0.0005	95.						
****	6006	0.100	-0.0005	110.-0.003	120.-0.012	140.-0.012	200.					
****	* HEAT GENERATION VERSUS TIME (NOT USED)											
****	6011	1.0	0.0	1.0	0.1	1.14	75.213	1.15	1.2.5	0.66	6.056.15.	
****	6012	0.45	50.039	100.0315	300.024	1000.						
****	* REACTIVITY VS. TIME											
****	6021	0.0	0.0	14.9	-0.08	15.-0.08	10000.					
****	* DELAYED NEUTRON FRACTIONS											
****	6031	.226-3.	1554-2.	1418-2.	2954-2.	816-3.	.339-3					

Table 2.26-3. (Cont'd)

	* PRECURSOR DECAY CONSTANTS							
6041	.125	-1	.3080	-1	.1140	.307	1200+1	.3210+1
	* INITIAL PRECURSOR CONCENTRATIONS							
6051	1.0	1.0	1.0	1.0	1.0	1.0	1.0	1.0
	* FISSION PRODUCT POWER FRACTIONS							
6061	.0183	.0161	.017	.0165	.06	-.076	.0423	-.048
	* DECAY CONSTANTS OF FISSION PRODUCTS							
6071	3.212	-5	7.996	-4	.0050849	.044887	.31801	.43779
6072	1.2104	5.2543	8.5752	0.	0.	0.	0.	0.
	* NORMALIZED CONCENTRATION OF FISSION PRODUCTS							
6081	1.	1.	1.	1.	1.	1.	1.	1.
	*STEAM GENERATOR							
7001	21	251785.	536.4					
7002	22	21422.9	591.					
7003	23	251785.	536.4					
7004	24	21422.9	591.					
7005	74	51970.3	568.4					
7006	89	503359.5	536.					
7007	90	488230.1	536.2					
7008	91	51970.3	568.4					
7009	92	503359.5	536.					
7010	93	488230.1	536.2					
7011	0.0	10000.	1.0	10000.	0.5	44.44		
7021	1.0	0.0	1.0	100000.				
7100	1	1						
	*STEAM GENERATOR LOGIC PARAMETERS							
7201	1200.0	470.5	36.0					
	*MAIN FEEDWATER COASTDOWN							
7211	2305.55	0.0	0.0	0.17	0.0	0.0	600000.0	
	*ALXILIARY FEEDWATER VERSUS SECONDARY PRESSURE							
7231	182.72	0.	182.72	872.87	138.59	1260.	68.95	1527.5
	*RELIEF VALVE ACTUATION VERSUS TIME (AFTER SCRAM)							
7251	1200.	0.	1200.	1.	5			
	*RELIEF VALVE FLOW VERSUS SECONDARY PRESSURE							
7271	0.	0.	1199.999	650.83	1200.	650.83	1260.	1533.89
7272	1533.89	1500.						
	*SAFETY VALVE FLOW VERSUS SECONDARY PRESSURE							
7291	0.	0.	1249.999	1242.5	1250.	1242.5	1270.	2480.
7292	2480.	1290.	3757.78	1290.001	3757.78	1300.		
	* PUMP CHARACTERISTICS							
8001	1	2.73	2.37	2.07	1.85	1.69	1.51	1.40
8002	2	.77	.50	.29	.14	.05	-.02	-.10
8003	3	-2.44	-2.30	-2.16	-1.93	-1.69	-1.45	-1.09
8004	4	.71	.69	.74	.83	.94	1.06	1.21
8005	5	1.96	1.56	1.25	1.13	.93	.77	.74
8006	6	.49	.16	-.18	-.38	-.53	-.87	-1.29
8007	7	-4.73	-4.00	-3.31	-2.78	-2.23	-1.64	-1.0
8008	8	.41	.60	.78	.91	1.02	1.13	1.21
	* PUMP ELECTRIC TORQUE VS. ACTUAL PUMP SPEED							
8016	38	0.0	39624.	120.	41626.	240.	44427.	360.
8017	52832.	600.	60437.	720.	70043.	840.	82851.	960.
8018	90456.	1128.	40025.	1181.	0.	1200.	-40025.	1219.
8019	-90456.	1320.	-82851.	1440.	-70043.	1560.	-60437.	1680.
8020	-52832.	1800.	-48030.	1920.	-44427.	2040.	-41626.	2160.
	+ -39624., 2280., -38023., 2400., -39624., 2520., -41626., 2640.							
	+ -44427., 2760., -48030., 2880., -52832., 3000., -60437., 3120.							
	+ -70043., 3240., -82851., 3360., -90456., 3480., -90456., 3528.							
	+ -40025., 3581., 0., 3600.							

Table 2-26-3. (Cont'd)

* PUMP SHUTDOWN PARAMETERS										
word1	8021	0	1.0	0	0	0	0	0	0	0
word2	8022	0	1.0	0	0	0	0	0	0	0
word3	8023	0	1.0	0	0	0	0	0	0	0
word4	8024	0	1.0	0	0	0	0	0	0	0
* PUMP PARAMETERS										
word1	8031	44	314.82	36300	1187.16	87763	4.28	1187.15		
word2	8032	45	314.82	36300	1187.16	87763	4.28	1187.15		
word3	8033	46	314.82	36300	1187.16	87763	4.28	1187.15		
word4	8034	47	314.82	36300	1187.16	87763	4.28	1187.15		
* PUMP MOTOR CONSTANTS										
word1	8041	71	111.429	54	35870	117.54	1.732	74.048	0.48690	390.115
word2	8042	0	0.00102	0.43320	1925	72700				
word3	M1	8051	-26.6026	50.9645	-33.6919	9.3302				

INPUT FILE OCCUPIES 5606 DATA WORDS

INPUT COMPLETE

Table 2.26-4. Microfiche Plots for Runs Listed in Table 2.26-1



3.1. In reference to Table 4.2-2, how are the break opening times determined for the 2A cold leg break? (Section 4.4)

Response

Generic BOT versus BOA curves for skirt- and nozzle-supported plants were used to determine the break opening time for each specific break opening area. Please refer to the response to question 2.6, Figures 2.6-1 through 2.6-4.

3.2. The comparison indicated in paragraph 4 between Figures D-6 and D-7 is unclear. Explain. (Section 5.1)

Response

Paragraph 4 of reference 2 should have stated:

By comparing Figure D-4 with the lumped-mass model shown in Figure D-5, a correlation can be seen between the components and the model elements representing them. The model-component correlation is addressed in more detail in Appendix D.



3.3. Provide the basis for the selection of the composite elements. (Section 5.1.4)

Response

One equivalent element represents all CRDMs. Section properties of one CRDM are used to determine the mass and stiffness of one drive unit. Mass and stiffness are then multiplied by the total number (no shear coupling) of CRDMs to define one equivalent element.

3.4. What method was used to calculate the stiffness values? (Section 5.1.4)

Response

Stiffness values were calculated for one CRDM and then multiplied by the total number of drives. Actual CRDM dimensions were used to calculate the properties of a single CRDM. Each CRDM unit is supported so that no shear connection exists between the drives.

3.5. Was pipe whip also considered for breaks outside the RPV cavity? (Section 5.2)

Response

Yes. Data presented in Tables 9.8-1 through 9.8-7 of reference 2 are summaries of pipe whip restraint loads for the various plants. Hot and cold leg guillotine breaks at the elbow referred to in these tables are for breaks in the steam generator cavity and are shown in Figure 4.1-1 of reference 2.

- 3.6. Expand the description of the initial conditions used in this analysis. Were the steady-state momentum forces considered in determining these initial conditions? (Section 5.2.1)

Response

The initial conditions for the pipe whip analysis were applied so that the state of stress in the piping to be analyzed reflects the normal operating conditions of the plant.

The thermal conditions were reflected by applying the temperature and coefficient of thermal expansion of the isolated piping model. Displacements were applied to the boundary conditions to reflect the growth of the balance of plant and its effect on the isolated model. Displacements were taken from existing stress reports and associated analyses. Since all skirt-supported plants are dimensionally similar, Rancho Seco displacements were applied to all of these plants. Thermal displacements for Davis Besse 1 were taken from the original Davis Besse 1 analysis.

The weight effects on the piping were applied using techniques similar to those used to apply thermal effects. The deadweight of the isolated pipe was applied while boundary condition displacements were superimposed with the thermals. Original Rancho Seco and Davis Besse 1 analyses were the source of boundary displacements due to deadweight.

The steady-state fluid hydraulics were reflected by applying the operating pressure to the piping. The momentum effects were found to be less than 5% of the steady-state hydraulics, which includes the pressure stress. Therefore, momentum effects were not included.

An investigation was conducted to determine the pipe whip sensitivity to the operating stress since plant-unique operating stresses were not imposed. For this investigation, pipe whip was run with and without operating stresses. The resultant pipe displacements and restraint loads were within 10% of each other. Thus, it was concluded that exact reflection of operating stress was unnecessary.

3.7. Discuss in detail the full loop model used to determine the operating condition loads. (Section 5.2.2.7)

Response

The loop model used for boundary displacements for the skirt-supported plants was the Rancho Seco model, which was used for contract analysis in support of the Rancho Seco FSAR.<sup>11</sup> For the nozzle-supported plant, boundary displacements were taken from the Davis Besse 1 contract analysis, which supports the Davis Besse 1 FSAR.<sup>10</sup> A general description of modeling techniques for these loop models is presented in sections 4, 5, and 12 of topical report BAW-10131.<sup>12</sup>

3.8. Provide more detail on the iteration procedure (include break area and break time) described in Figure 5.2-1. (Section 5.2.2.9)

#### Response

Initially the plant is operated at steady-state flow, temperature, and pressure with the piping system intact. An instantaneous pipe severance is postulated to occur, which results in an unbalanced force on the piping that had, prior to the break, acted as an axial load (longitudinal pressure stress  $P r_m / 2T$ ) in the continuous piping system. This unbalanced force is equal to the pressure in the first elbow adjacent to the break times the flow area inside the pipe ( $P \cdot A_{\text{pipe}}$ ). This primary force in a pipe whip analysis causes the majority of the motion in the broken piping. As the pressure in the adjacent elbow decays, the force driving the broken pipe decays. As the pipe opens faster, the driving force decays faster. To avoid allowing this force to decay too rapidly, a "zero area" break is hydraulically hypothesized (hydraulic analysis No. 1). This is accomplished by assuming that the break causes no decay in the driving force and thus remains at a constant  $P \cdot A_{\text{pipe}}$ . This method results in a pipe whip analysis (pipe whip analysis No. 1) with high restraint loads accompanied by pipe motion at a faster opening rate than normal, which is conservative. If the loads are beyond allowable or the opening area/rate is too conservative for the unbroken loop, so that loading problems appear elsewhere, the hydraulics are iterated again in hydraulic pressure analysis No. 2 using the opening areas and rates predicted by the zero area pipe whip No. 1 analysis.

These iterated forces (hydraulic pressure analysis No. 2) are used as input to pipe whip analysis No. 2 and new restraint loads and opening rates are determined.

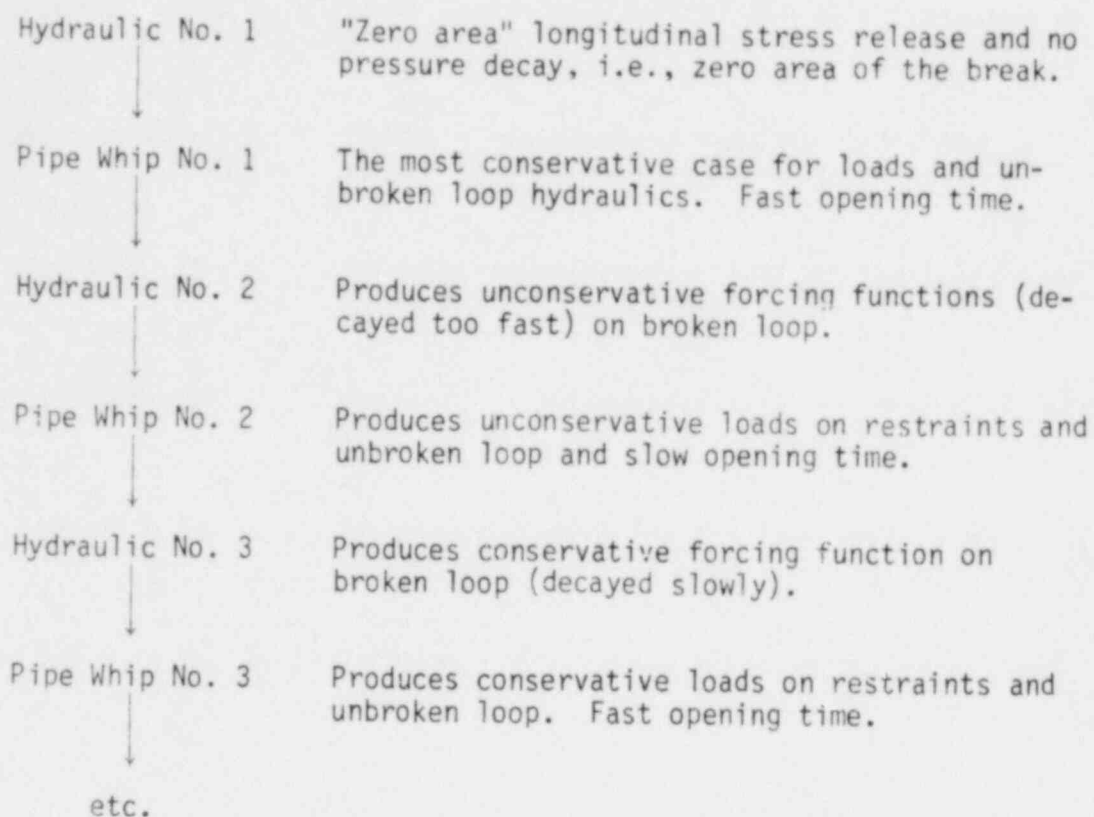
Loads from this iteration are lower, and opening rates are slower than those from pipe whip No. 1. Since the hydraulics for this iteration (hydraulics No. 2) have the fastest opening rates (a zero area) possible for the restraint and piping scheme, the pressures decay as rapidly as possible. The new opening rates from this iteration, which are slower than pipe whip No. 1, must then be addressed.

Opening rates from pipe whip No. 2 are the slowest possible for the piping and restraint configuration since the input forces from hydraulics No. 2 decay

the fastest as a result of the "zero area" concept. Thus, the hydraulics generated by hydraulics No. 3 are conservative for the driving forces using opening rates from pipe whip No. 2; yet they are converging from the zero area concept.

Next, pipe whip No. 3 is run using the hydraulics transient results (pressure) and opening rates from hydraulics No. 3 and pipe whip No. 2 analyses, respectively. Since the input is conservative, the results from pipe whip No. 3 are conservative. This is usually the last iteration, and these results are reported for restraint and pipe whip. These restraint loads are conservative and opening rates are fast, which is conservative for the component analysis and the unbroken loop analysis. Figure 3.8-1 is a flow chart for this iterative sequence.

Figure 3.8-1. Pipe Reaction Force Iterative Process





3.9. Clarify this discussion. (Section 5.3.6.3)

Response

Given the horizontal and vertical pipe reactions H and V on the LOCA ring, the friction force was determined by the formula

$$F = \rho(H^2 + V^2)^{\frac{1}{2}}$$

where the friction coefficient  $\rho = 0.42$ .

This friction force acts parallel to the pipe axis. A conservative horizontal force was then calculated by the formula

$$R = (H^2 + F^2)^{\frac{1}{2}}.$$

This force R was applied in the horizontal direction in the plane of the LOCA ring with the vertical force V to conservatively determine the concrete stresses around the ring.

3.10. Was a scram time evaluation performed? (Section 6)

Response

A scram time evaluation was not performed as a part of the Phase II analysis, and we know of no licensing requirement for a post-LOCA trip evaluation.

The integrity of the control rod drive pressure boundary was demonstrated in the Phase II analysis; therefore, in our judgement, the drives would be capable of inserting the control rods after LOCA.

3.11. Supply the supporting data for the compressive strength value of embedment concrete used for the Rancho Seco plant. (Section 6.2.1.4)

Response

The compressive strength values for the embedment concrete are based on 90-day strength tests. Where three strength tests were available, the 90-day strength was taken as either the average of the three test strengths or the minimum test strength plus 500 lb/in.<sup>2</sup>, whichever was the lesser. This is consistent with ACI 214-77<sup>15</sup>.

One strength test consists of two cylinder tests. For the Rancho Seco plant, only two series of strength tests (four cylinder tests) were available<sup>16</sup>. The results of these tests are given in Table 3.11-1. The average of the 90-day strength tests, 7275 lb/in.<sup>2</sup>, was taken as the in situ concrete strength.

Table 3.11-1. Compressive Strength of Embedment Concrete for Rancho Seco

Test No.	Tested strength, lb/in.		
	7-day	28-day	90-day
28	4492	6512	7340
29	4628	6237	7216
Avg	4560	6575	7275

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Response

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- 3.12. Supply a sample calculation demonstrating the procedure outlined for combining seismic and LOCA responses for the embedments. (Section 6.2.1.9)

Response

The following example calculation is provided for the Oconee plant embedment. The methodology used is that described in section 6.2.1.9 of reference 2.

The design seismic overturning moment on the reactor vessel support embedment is<sup>17</sup>

$$M_{eq} = 100 \times 10^3 \text{ kips in.}$$

This overturning moment was calculated using the pre-liftoff embedment condition. Furthermore, this moment is considerably less than that required to initiate liftoff.

The rotational tangent stiffnesses of the Oconee embedment before and after liftoff,  $K_1$  and  $K_2$ , are

$$K_1 = 6.875 \times 10^9 \text{ kips in./rad}$$

and

$$K_2 = 0.997 \times 10^9 \text{ kips in./rad.}$$

The rotational secant stiffness at the moment corresponding to the maximum LOCA loading is

$$K_3 = 1.534 \times 10^9 \text{ kips in./rad.}$$

Assume that the seismic overturning moment is governed by the post-liftoff rather than pre-liftoff embedment stiffness. The increase in  $M_{eq}$  can be determined as follows from the fixed-shape spectra given in NRC Regulatory Guide 1.60.<sup>18</sup>

The ratio of the post- and pre-liftoff tangent stiffnesses,  $R_K$ , is

$$R_K = 0.15.$$

Thus, the ratio of the post- and pre-liftoff fundamental periods of the RV on the embedment,  $T_K$ , is

$$T_K = 2.63.$$

Assuming damping equal to 10% of critical (for highly stressed reinforced concrete), the maximum increase in response due to the increase in fundamental period,  $A_K$ , is

$$\begin{aligned} A_K &= e^{0.494 \ln T_K} \\ &= 1.61. \end{aligned}$$

Thus, the upper-bound seismic overturning moment is

$$\begin{aligned} M'_{eq} &= 1.61 \times M_{eq} \\ &= 161 \times 10^3 \text{ kips in.} \end{aligned}$$

The energy demand associated with this loading,  $E_{eq}$ , is

$$\begin{aligned} E_{eq} &= \frac{(M'_{eq})^2}{2 \times K_3} \\ &= 8.45 \text{ kips in./rad.} \end{aligned}$$

The peak LOCA overturning moment for Oconee is  $M_L$ , where

$$M_L = 2496 \times 10^3 \text{ kips in.}$$

This moment produces a rotation of  $\theta_L$ , where

$$\theta_L = 1.630 \times 10^{-3} \text{ rad.}$$

The increased rotation,  $\theta_{L+eq}$ , required for the seismic energy demand,  $E_{eq}$ , to be met is, from the moment-rotation curve for Oconee,

$$\theta_{L+eq} = 1.633 \times 10^{-3} \text{ rad.}$$

This corresponds to a total seismic-plus-LOCA overturning moment of

$$M_{L+eq} = 2501 \times 10^3 \text{ kips in.}$$

As an alternative, the total seismic-plus-LOCA overturning moment was also calculated using the SRSS (square root sum of the squares) method, i.e.,

$$\begin{aligned} M_{L+eq} &= [(2496 \times 10^3)^2 + (100 \times 10^3)^2]^{\frac{1}{2}} \\ &= 2498 \times 10^3 \text{ kips in.} \end{aligned}$$

Similar calculations were performed for the remaining plants.

3.13. Was a core barrel shell analysis performed: (Section 6.5)

Response

A core barrel analysis was performed. The results of the evaluation are included in the response to question 3.24.

3.14. Essentially four methods of demonstrating acceptability of the core flood line piping are discussed in this section: (Section 6.9.3)

1. Response spectrum.
2. Response spectrum removing failed support.
3. Time history.
4. Time history removing failed supports.

Methods 1 and 3 are acceptable with no further justifications; however, methods 2 and 4 require additional quantitative justification.

#### Response

The core flood line piping for B&W plants feeds directly into the reactor vessel. After a hot or cold leg guillotine break at the reactor vessel nozzle, the reactor vessel rocks in response to the applied asymmetric loads. From the vessel motion, displacement, velocity, and acceleration, time history motions were calculated at the core flood line nozzle attachment points.

The core flood lines for all B&W plants were analyzed using the response spectrum (methods 1 and 2) or time history (methods 3 and 4) approach. The results of these linear elastic analyses indicate that the following plants have overloaded core flood line supports:

Three Mile Island 1 -	Two snubbers loop A Two snubbers loop B
Crystal River 3 -	One snubber loop A Three snubbers loop B
Arkansas Nuclear One, Unit 1 -	One X-stop loop A One X-stop loop B

Table 3.14-1 summarizes the supports that are overloaded due to the LOCA event.

The function of the core flood lines after the LOCA event is to inject water into the reactor vessel. For this to occur, the piping and valves must remain functional. To show functionality, the pipe stress values were evaluated to service level C, and the valve stresses were evaluated to service level B.

Under methods 1 and 3, (assuming no support failures), all core flood lines have primary stresses below level C limits. All core flood lines have valve stresses below level B limits. All core flood lines have secondary stress limits below the  $3S_m$  limit except TMI-1 loop A, which is only 14% over.



Under methods 2 and 4, (assuming failed supports removed), the secondary stress on TMI-1 loop A is below the  $3S_m$  limit. This result is reasonable since secondary stresses are due to imposed anchor displacements. With the yielding or failure of the TMI-1 loop A snubber, the piping becomes more flexible, allowing for easier acceptance of the imposed anchor movement.

Under methods 2 and 4, (assuming failed supports removed), all core flood lines have primary stresses below level C limits, valve stresses below level B limits, and secondary stresses below the  $3S_m$  limit. We can firmly conclude that all core flood lines would remain functional during the LOCA event if the identified supports were nonexistent.

The actual phenomena of support failures cause the subject core flood lines to behave in a nonlinear fashion. Engineering judgment based on experience with nonlinear systems indicates that secondary stress values in the piping will decrease due to support yielding. Analyses using methods 2 and 4 show that primary and valve stresses remain acceptable if the subject failed supports were not in place at the start of the event.

Therefore, it is concluded that all core flood lines should remain functional during the LOCA event. Further data are given in Tables 3.14-2 through 3.14-4.

Table 3.14-1. Failed Supports on Core Flood Lines

<u>Loop</u>	<u>Support</u>	<u>Description</u>
<u>TMI-1</u>		
A	CF-6	2.5-in. bore, 5-in. stroke snubber
A	CF-7	
B	PA-9	
B	PB-10	
<u>CR-3</u>		
A	CFH-16	2.5-in. bore, 5-in. stroke snubber
B	CFH-14	
B	CFH-13	
B	CFH-12	
<u>ANO-1</u>		
A	DH-196	WF 4 x 13 x-stop
B	DH-164	

Table 3.14-2. Support Reaction Data

<u>Support</u>	<u>P+G+LOCA reaction, lb</u>	<u>P+G+LOCA ratio to one-time load rating</u>
<u>TMI-1</u>		
CF-6	43,738	1.70
CF-7	30,544	1.19
CF-8	22,600	0.88
CF-9	11,161	0.43
CF-14	25,461	0.99
CF-15	58,993	0.89
PA-9	49,922	1.94
PB-10	55,856	2.17
<u>ANO-1</u>		
D196	52,552	2.43
D198	23,074	0.60
<u>CR-3</u>		
CFH-12	37,698	1.80
CFH-13	31,113	1.48
CFH-14	54,429	2.60
CFH-15	38,859	0.67
CFH-16	33,550	1.60
CFH-17	15,362	0.26
CFH-18	18,139	0.86
CFH-19	31,227	0.54

Table 3.14-3. Core Flood Line Valve  
Stress Ratios

Valve type & size, in.	Node	Stress ratio to level B limit	
		With supports	Without supports
<u>TMI-1</u>			
14 check	A-19	0.41	0.62
14 check	A-28	0.66	0.60
14 check	B-39	0.44	0.48
<u>CR-3</u>			
14 check	A-23	0.44	0.54
14 check	B-63	0.54	0.51
<u>ANO-1</u>			
14 check	A-250	0.27	0.30
14 gate	A-345	0.55	0.58

Table 3.14-4. Core Flood Line Peak Pipe Stress Data

Case	Location	Stress, psi		Ratio
		Primary	Allowable	
<u>TMI-1, Loop A</u>				
CL,W	Nozzle	22,050	31,280	0.71
CL,W0	Nozzle	22,239	31,280	0.71
HL,W	Nozzle	14,350	31,280	0.46
HL,W0	Nozzle	26,774	31,280	0.86
HL,W	Elbow	13,014	31,280	0.42
HL,W0	Elbow	23,951	31,280	0.77
CL,W	Branch	49,417	51,750	0.96
CL,W0	Branch	44,443	51,750	0.86
<u>TMI-1, Loop B</u>				
HL,W	Nozzle	18,894	32,327	0.59
HL,W0	Nozzle	25,629	32,327	0.79
<u>CR-3, Loop B</u>				
HL,W	Nozzle	24,207	31,395	0.77
HL,W0	Nozzle	21,224	31,395	0.68
<u>ANO-1, Loop A</u>				
HL,W	Elbow	13,867	32,378	0.43
HL,W0	Elbow	15,784	32,378	0.49

Legend - CL: cold leg break  
 HL: hot leg break  
 W: with supports  
 W0: with failed supports removed

3.15. How was the 7% critical damping incorporated in the dynamic analysis?  
Specify which modes were critically damped. (Section 6.10.4.5)

Response

Damping for the dynamic analysis was specified through Rayleigh ( $\alpha$ ,  $\beta$ ) damping. The mass and stiffness proportional factors  $\alpha$  and  $\beta$  were determined by setting 7% damping at frequencies of 40 and 150 Hz ( $\alpha = 27.37$ ,  $\beta = 1.1727 \times 10^{-4}$ ).

3.16. Demonstrate with a detailed calculation that the dynamic load factor is 1.15. (Section 6.10.4.6)

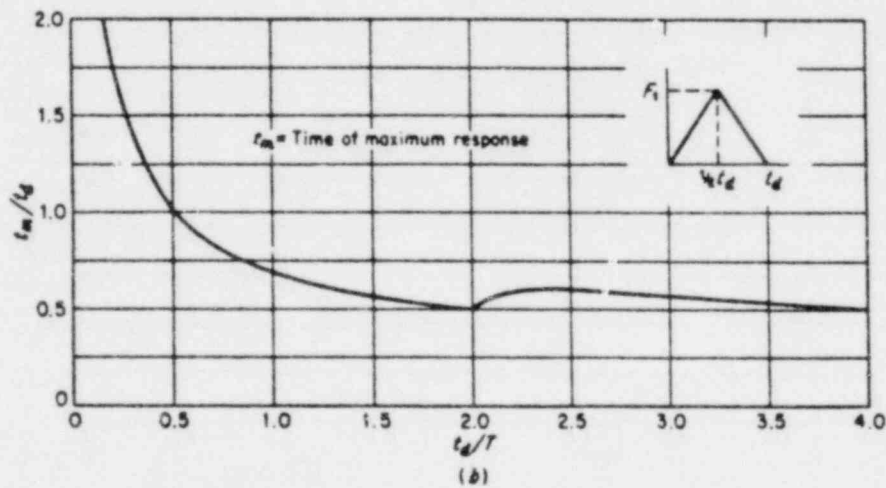
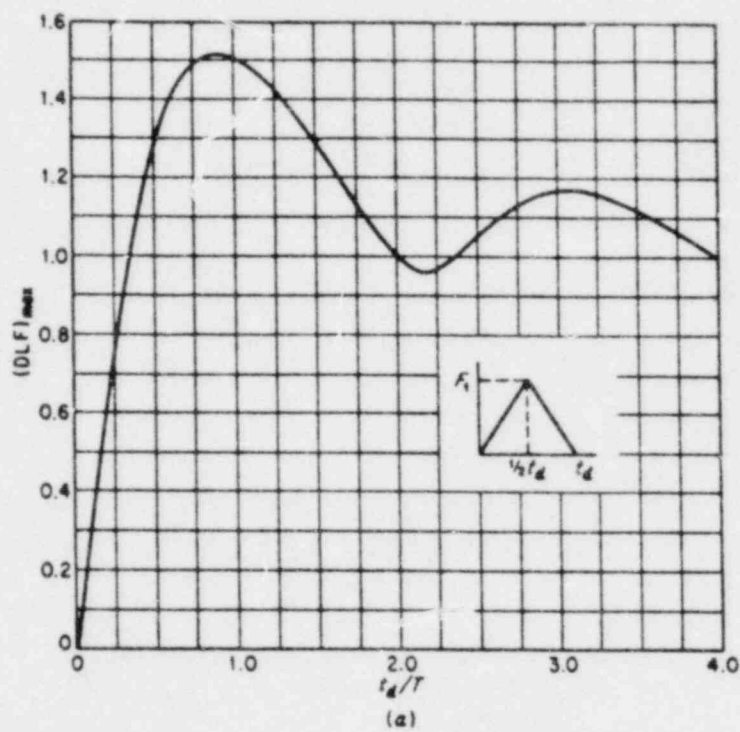
Response

The loads on the cavity walls can be approximated as a triangular pulse with pulse duration from 0.06 to 0.08 s (Figures 9.10-1 through 9.10-6<sup>2</sup>). The range of the first mode frequency of cavity walls is from 30 Hz for ANO-1 (orthotropic analysis) to 50 Hz for Rancho Seco.

Considering the range of pulse durations  $t_d$  and the range of frequencies, a range of ratios  $t_d/T$  can be derived. In Figure 3.16-1 the dynamic load factor (DLF) for a triangular load pulse is given for values of  $t_d/T$ .<sup>19</sup> For the range of interest,  $1.82 \leq t_d/T \leq 4.0$ , the maximum DLF is about 1.15.

Dynamic and static analyses were run for both the Arkansas and Rancho Seco cavity walls. Stress results from both analyses were compared. Actual peak dynamic stress results were less than those predicted by the DLF curve.

Figure 3.16-1. Maximum Response of One-Degree Elastic Systems (Undamped) Subjected to Isosceles Triangular Load Pulse



3.17. Demonstrate that the strain rates are sufficiently high to justify an increase in yield stress of 10%. (Section 7.1.1.1)

Response

The strain rates for the rupture restraints near yield stress range from 0.26 to 1.97 in./in.-s as given in Table 3.17-1. For these values of strain rate, experimental data show 25% increase in yield stress.<sup>20</sup>

Standard Review Plan 3.6.2 suggests a 10% increase in yield stress for rupture load analyses.<sup>21</sup>

Therefore, using a 10% increase in yield stress is conservative.

Table 3.17-1. Rupture Restraint Strain Rate

<u>Rupture restraint</u>	<u>Strain rate, in./in.-s</u>
Ocone hot leg bumper restraint	Stresses below yield
Ocone hot leg collar restraint	Stresses below yield
TMI-1 U-bar rupture restraint	1.13
TMI-2 U-bar rupture restraint	0.99
CR hot leg U-bar rupture restraint	0.26
ANO-1 hot leg strap rupture restraint	0.61
Rancho Seco hot leg upper rupture restraint	Stresses below yield
Rancho Seco hot leg lower rupture restraint	1.97



3.18. Explain the shear and tension interaction equation development. (Section 7.1.1.4)

Response

From pages 35 and 54 of reference 22, the interaction formulas based on the experimental data for combined tension and shear for steel are

$$\frac{fv^2}{(0.75 S_y)^2} + \frac{ft^2}{S_y^2} \leq 1 \quad \text{for a rivet}$$

and

$$\frac{fv^2}{(0.6 S_y)^2} + \frac{ft^2}{S_y^2} \leq 1 \quad \text{for a bolt.}$$

The interaction formula used for rupture restraint qualification was conservatively taken as:

$$\frac{fv^2}{(0.577 S_y)^2} + \frac{ft^2}{S_y^2} \leq 1$$

for both pins and bolts. Hence, the interaction equation for shear and tension conservatively models experimental data.

- 3.19. Demonstrate that the strain rates are sufficiently high to justify an increase in concrete strength of 25% in compression and 10% in shear. (Section 7.2.1)

Response

From Appendix C of ACI 349-76<sup>2,3</sup>, impulse loads are defined as including blast pressurization and compartment pressurization. For these loadings, a dynamic increase factor (DIF) may be applied to static material strengths. Appendix C gives DIFs of 1.25 for concrete under axial and flexural compression, and 1.10 for concrete under shear.

For the reactor vessel embedment, the concrete strain rate during the impulsive stage of the loading is approximately 0.067 in./in.-s. This value was determined by analysis. For strain rates of this order, Park and Paulay<sup>2,4</sup> give the DIF as approximately 1.25, which confirms the applicability of the ACI Code.<sup>2,3</sup>

3.20. Justify the use of level D stress limits for tees and branch connections. (Section 7.3.2.1)

Response

Appendix A of the proposed revision to Standard Review Plan 3.9.3 (PSRP 3.9.3<sup>25</sup>) states:

2.3.4. Service Limit D

The use of a service limit not greater than Level D for Class 1, 2, and 3 components of Type II safety-related systems and component supports of Type I and II safety-related systems, in lieu of service limit B, when subjected to the loadings described in section 2.3.2(b) (2) may be permitted for designated components and component supports provided that (1) the operability of active pumps and valves is demonstrated at service limit D (see section 3.1); (2) the functional capability requirements of section 3.3 at service limit D are satisfied; and (3) stresses produced by constraint of free-end displacement and anchor point motions satisfy section 3.4 for the above loadings.

3.3.2. Piping Components for Which Functional Capability is Ensured Without Further Demonstration

Functional capability may be considered ensured without further demonstration for Class 1, 2, and 3 piping components with an outside diameter-to-thickness ratio not greater than 50 ( $D_o/t \leq 50$ ) which meet one of the following criteria:

a. Class 1 piping components:

1. All piping components for which Service Limit C is permitted by section 2.3.3, provided the primary stress intensity as calculated by equation 9 of NB-3652<sup>2/</sup> does not exceed  $1.5 S_y$ .
2. Tees and branch connections for which Service Limit D is permitted by section 2.3.4, provided the primary stress intensity as calculated by equation 9 of NB-3652<sup>2/</sup> does not exceed  $2.0 S_y$ .

The  $D_o/t$  ratios for all core flood lines was less than 50. The peak check valve stress on the core flood lines was less than service level B limits. Hence, the use of service level D limits for tees and branch connections is justified.

3.21. Specify the BOA time considered in determining the loadings shown in Table 8.1-1. (Section 8.1)

Response

Break opening times considered in determining the loadings shown in Table 8.1-1 of reference 2 are tabulated in the response to question 2.6, Tables 2.6-1 and 2.6-3.

3.22. The vertical load shown in this and subsequent figures does not appear to have reached its maximum value. Please discuss in detail. (Section 8.1)

Response

With reference to the vertical load curves shown in section 8.1 of reference 2, it was determined that these loads peak after the 0.5-second time shown. This can be seen by referring to Figures 8.1-1g, -2g, -3h, and -4h of reference 2. The data in Tables 9.1-2 through 9.1-5<sup>2</sup> were used to analyze the supports. These data show that the moments are very large in comparison to the magnitude of the vertical forces. This is demonstrated by considering the following stress calculation on the reactor vessel support skirt:

Support skirt diameter, in.	179.5
Support skirt thickness, in.	2.0
Number of holes in skirt,	12
Diameter of holes in skirt, in.	9.25
Vertical force on TMI-1/CR-3 from Table 9.1-3, kips	6358.2

$$\text{Stress: } \frac{P}{A} = \frac{6,358,200}{(179.5\pi - 12 \times 9.25)\pi} = 7,019 \text{ psi}$$

Even if vertical loads in Tables 9.1-3 through 9.1-5<sup>2</sup> were increased by 50%, the stresses from these loads would not approach the stress due to the moment loading.

3.23. Identify the break locations from Figure 4.1-1 associated with each of the loadings tabulated in Table 8.1-1. (Section 8.1)

Response

With reference to Figure 4.1-1 of reference 2, break locations would be at the terminal ends of the reactor vessel nozzles for the specified hot or cold leg break. Data referenced in Table 8.1-1 for Davis Besse 1 (nozzle supported plant) were for the actual inlet and outlet nozzle break areas. A spectrum of break sizes - 2.0A, 1.5A, 1.0A, 0.6A, and 0.3A - was used for all other plants referenced in Table 8.1-1.<sup>2</sup> Additional data are provided in response to question 2.6.

3.24. Supply quantitative results and allowable values. (Section 10.5)

Response

The quantitative results for the reactor internals/core support assembly are found in Tables 3.24-1 through 3.24-5.

Table 3.24-1. Core Support Shield Stress Analysis  
Results for Faulted Conditions

<u>Location</u>	<u>Stress category</u>	<u>Maximum allow. stress, psi</u>	<u>Max calculated stress, psi</u>	
			<u>Skirt-supported RV</u>	<u>Nozzle-supported RV</u>
Core support shield	Pm	39,800	14,109	14,517
	Buckling	319	145.4	187.9
CSS/CB flanges and joint				
Bolted joint	Pm	91,000	28,460	35,381
CB upper flange	Pm	38,200	4,564	5,675
	Bearing	35,400	8,439	10,492
	Shear	19,080	4,974	6,183
CSS lower flange	Pm	38,200	8,473	9,719
	Pm + Pb	57,300	19,391	22,242
	Bearing	35,400	8,345	9,572
CSS upper flange	Limit load, kips	$1.98 \times 10^4$	$1.82 \times 10^4$	$1.68 \times 10^4$

Table 3.24-2. Lower Grid Assembly Stress Analysis  
Results for Faulted Conditions

Location	Stress category	Maximum allow. stress, psi	Max calculated stress, psi	
			Skirt-supported RV	Nozzle-supported RV
<u>LGA Shells</u>				
CB/LGA bolts	Pm	86,600	18,712	21,464
	Stripping	19,080	6,374	3,735
	Bearing	35,400	4,321	4,957
CB lower flange	Pm	38,160	4,080	4,680
<u>Lower Grid Assembly</u>				
Grid pad	Bearing	35,400	4,921	3,281
Grid pad/rib sec joint - bolt	Shear	38,160	10,338	7,419
	Pm	19,080	3,465	2,486
Rib section	Pm	38,160	20,669	14,515
	Pm + Pb	57,420	33,900	26,653
Support post/rib sec joint - bolt	Pm	38,160	3,399	2,266
Rib sec/shell forging joint - bolt	Pm	38,160	20,187	13,458
Support posts	Pm	38,160	8,686	6,443
FD plate	Pm	57,240	1,908	1,272
FD plate/shell forging joint - weld	Pm	22,896	2,159	1,439
Support post/FD plate welded joint	Pm	15,264	560	373
Support forging	Pm	38,160	1,712	1,425
	Pm + Pb	57,240	15,194	12,652
Support post/support forging joint - weld	Pm	30,528	16,686	23,758
Shell forging	Pm	38,160	885	779



Table 3.24-3. Flow Distributor Assembly Stress Analysis  
Results for Faulted Conditions

Location	Stress category	Maximum allow. stress, psi	Max calculated stress, psi	
			Skirt-supported RV	Nozzle-supported RV
Flow distributor head/LGA joint				
FDH/LGA bolts	Pm	91,000	11,340	11,340
	Tear-out	20,256	184	184
	Bearing	37,560	1,456	1,456
FDA shell	Pm	40,512	125.7	125.7
	Pm + Pb	60,768	1,125.4	1,125.4
FDA flange	Pm	40,512	284.6	284.6
	Pm + Pb	60,768	6,240	6,240

Table 3.24-4. Core Basket Assembly Stress Analysis Results for Faulted Conditions

Location	Stress category	Maximum allow. stress, psi	Max calculated stress, psi	
			Skirt-supported RV	Nozzle-supported RV
Core barrel	Pm	38,400	10,521	10,826
	Buckling	20,750	8,572	8,583
Core barrel assembly				
Baffle-former A-A bolts	Pm	38,160	8,082	6,510
Barrel-former A-A bolts	Pm	38,160	22,503	16,135
Thermal shield				
Lower end	Pm	38,160	1,525	1,824
Upper end	Pm	38,160	10,316	10,711
TB/LG shell forging bolts	Pm	91,000	43,969	40,451
TS upper restraint	Bearing	35,400	5,999	6,234
	Shear	19,080	8,145	8,464
Bolts	Pm	91,000	26,849	27,902
Dowels	Shear	51,960	3,988	4,144

Table 3.24-5. Plenum Assembly Stress Analysis Results  
for Faulted Conditions

Location	Stress category	Maximum allow. stress, psi	Max calculated stress, psi	
			Skirt-supported RV	Nozzle-supported RV
Plenum cover	Bearing	27,300	14,742	17,059
Plenum cylinder	Pm	39,120	27,534	24,613
CRGT	Buckling	21,500	14,175	11,669
Slotted region	Pm	39,190	6,003	8,253
	Buckling	21,250	6,003	8,253
Perforated region	Pm	39,190	4,277	5,909
	Buckling	21,250	4,277	5,909
Lower joint - bolt - dowel	Pm	39,190	12,398	11,745
	Shear	51,960	3,408	4,797
CRGT/plenum cover joint	Pm	13,776	2,340	2,672
Upper grid assembly				
Rib section	Pm	39,245	16,903	10,775
	Pm + Pb	58,867	27,130	16,934
UG pad joint	Dowel	51,960	6,682	4,356
	Bolt	39,187	24,675	16,088

3.25. Provide a calculated load value which yields the spacer grid deflection indicated in Table 10.6-1 (section 10.6).

Response

Table 10.6-1 of reference 2 shows a spacer grid permanent deformation of 0.114 in. This deformation was determined as follows.

The calculated maximum impact load for the worst LOCA load case was 12,576 lb. Another impact of slightly less magnitude occurred before the maximum impact. The rest of the impacts were substantially less. These loads were determined at the hot operating reactor condition. An equivalent maximum impact load at room temperature was obtained by multiplying this load by the ratio of modulus of elasticity at room temperature to that at reactor operating temperature.

A curve of permanent deformation versus impact load was developed from spacer grid impact test data. A permanent deformation of 0.057 in. corresponds to the predicted maximum impact load. This deformation was doubled to 0.114 in. to account for the second impact. This is a conservative treatment of the experimental permanent deformation data to account for the multiple impacts resulting from the use of the nonlinear elastic plastic model of the spacer grid in the dynamic analyses. Spacer grid impact tests show that once the grid has plastically deformed, a second impact at about 90% of the buckling impact velocity results in much lower additional permanent deformation. Substantially lower impact loads (impact velocity = 50 to 80% of the buckling velocity) on the buckled grid do not cause additional permanent deformation.

3.26. Are all Owners Group plants supplied with Inconel grids? If not, explain. (Section 10.6)

Response

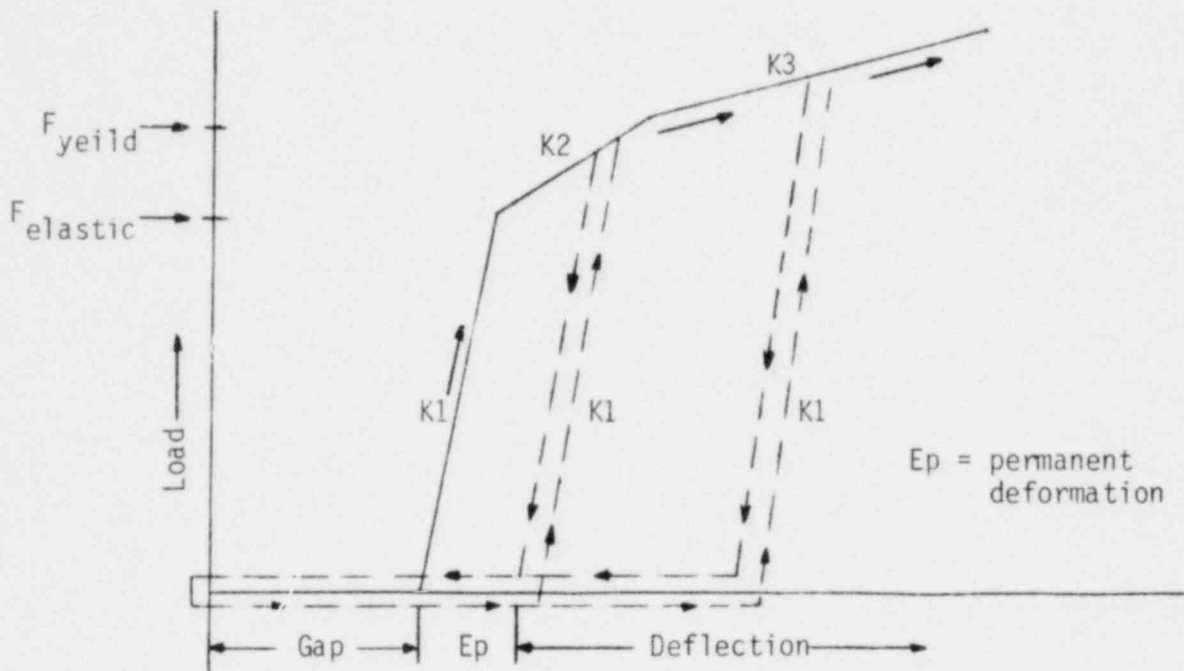
All Owners Group plants are supplied with Inconel grids except for demonstration assemblies that have been or soon will be inserted at Oconee. At present, Oconee 2 has one fuel assembly with Zircaloy intermediate spacer grids. Four demonstration fuel assemblies with Zircaloy grids are scheduled to go into Oconee 1.

The acceptable use of these demonstration assemblies has been documented in BAW-1565, Rev 1, and BAW-1533, Rev 1, for Oconee 2 and in BAW-1660 and BAW-1661 for Oconee 1.

3.27. Were nonlinear impact elements incorporated in the appropriate computer code to yield the spacer grid deflection? If so, were these elements nonlinear elastic or elastic plastic? Provide a discussion. (Section 10.6)

Response

Yes. Nonlinear elastic-plastic elements are incorporated in the STARS computer code. The spacer grid is represented as a trilinear gapped spring as shown below.



The spacer grid spring rates were determined from the room temperature spacer grid impact tests. Impact loads are calculated at hot operating conditions using this trilinear spring model. An equivalent load at room temperature was determined by multiplying this calculated load by a ratio of modulus of elasticity at room temperature to that at reactor operating temperature.

A permanent deformation was determined from a curve of permanent deformation versus impact load developed from the test data.

3.28. Are these results applicable for both broken and unbroken loop piping?  
(Section 10.7)

Response

The results in section 10.7 of reference 2 are for unbroken loop piping. For example, the moments and stresses given for a point on the A loop hot leg are produced by a postulated hot leg break in the B loop.

3.29. Identify which breaks yield which stresses. (Section 10.7)

Response

Tables 3.29-1 through 3.29-3 provide the following information: the plant having the largest load, the joint location at which the resultant moment acts, the LOCA case (break location) that produces the load, and the maximum resultant moment and actual bending stress at the joint.

Table 3.29-1. Additional Information for Table 10.7-1 of Reference 2, HLG LOCA Case<sup>a</sup>

<u>Pipe location, section</u>	<u>Plant</u>	<u>Joint No.</u>	<u>Moment <math>M_i</math>, ft-kips</u>	<u>Actual stress <math>f_b</math>, psi</u>
Hot leg, straight	TMI-1 CR-3	138	4,726.4	26,040
Cold leg, straight	TMI-2	89	2,150.1	24,622
Hot leg, elbow	TMI-2	158	3,097.3	44,967
Cold leg elbow	TMI-2	97	1,520.3	30,702
Cold leg, safe-end	TMI-2	149	1,338.4	15,349

<sup>a</sup>HLG: hot leg guillotine at reactor vessel outlet nozzle.



Table 3.29-2. Additional Information for Table 10.7-2 of Reference 2, Davis Besse 1 Plant, LOCA Case 1<sup>a</sup>

Pipe location, section	Joint No.	M <sub>i</sub> , ft-kips	Actual stress f <sub>b</sub> , psi
Hot leg, straight	74	2,673.4	18,822.3
Cold leg, straight	97	5,731.5	54,724.8
Hot leg, elbow	104	1,619.0	25,715.2
Cold leg, elbow	--	--	--
Cold leg, safe-end	142	2,880.9	25,142.8

<sup>a</sup>LOCA case 1: hot leg guillotine at reactor vessel outlet nozzle.

Table 3.29-3. Additional Information for Table 10.7-3 of Reference 2, Davis Besse 1 Plant, LOCA Case 1<sup>a</sup>

Elbow location	Joint No.	M <sub>i</sub> , ft-kips	Actual stress f <sub>b</sub> , psi
PIA1	110	5,458.95	66,090
	119	5,057.28	60,821
	135	3,239.37	40,469
	143	2,834.41	38,338
PIA2	107	5,511.61	63,066
	117	5,097.78	58,143
	133	3,244.01	38,448
	141	2,831.85	37,140
PIB1	106	5,521.31	65,073
	116	5,127.46	62,919
	132	3,279.88	43,585
	140	2,875.48	39,766
PIB2	108	5,480.16	62,654
	118	5,096.79	59,838
	134	3,279.11	41,144
	142	2,880.93	38,509

<sup>a</sup>LOCA case 1: hot leg guillotine at reactor vessel outlet nozzle.

3.30. Are the results in Tables 10.7-1 and 10.7-2 consistent with the equations in section 6.7? (Section 10.7)

Response

Yes. From section 6.7 of reference 2, the equation is

$$B_1 \frac{PD_o}{2t} + B_2 \frac{D_o}{2I} M_i \leq 3 S_m.$$

A sample calculation is performed to provide the results for the hot leg straight pipe in Table 10.7-1 of reference 2:

Hot Leg Straight Pipe

$$B_1 = 0.5 \quad D_o = 42.25 \text{ in.}$$

$$B_2 = 1.0 \quad t = 2.813 \text{ in.}$$

$$P = 2250 \text{ psi} \quad I = 68,107 \text{ in.}^4$$

$$M_i = 4726.4 \text{ ft-kips}$$

$$\begin{aligned} \text{Actual } f_b &= 0.5 \frac{(2250)42.25}{2(2.813)} + 1.0 \frac{42.25}{2(68,107)} 4726.4(12)1000 \\ &= 26,040.6 \text{ psi} \end{aligned}$$

3.31. Explain in detail the calculations performed to reflect the assumed 2A BOA. (Section 11.1.2)

Response

The response to question 3.33 addresses the 2A BOA.

3.32. What was the basis for assuming that all the skirt-supported plants were hydraulically similar to Midland 1 and 2? The same question applies for nozzle-supported plants. Are specific plant analyses planned in the future? (Section 11.1.2)

Response

The piping configurations for the 177-FA skirt-supported plants are very similar. Other plant parameters used to establish generic fluid conditions are tabulated in response to question 2.10 and are similar. Thus, it was concluded that the plants are hydraulically similar. Davis Besse 1 is the only operating 177-FA nozzle-supported plant; however, piping reaction forces were calculated using Davis Besse 2 and 3 plant data. These plants are hydraulically identical to Davis Besse 1. No additional analyses of asymmetric loading effects for pipe breaks in the steam generator compartment are planned.

Breaks in the RV compartment present the most severe challenge to core cooling, and extensive analysis has shown that the core flood line, RV internals, and the RV are qualified for core cooling. Review of available data for the steam generator compartment results in qualifying the remaining primary piping as maintaining the pressure boundary.

3.33. Justify multiplying the 1A break results by 2.0 to obtain 2A results. Why was a factor of 1.5 chosen instead of some other value? (Section 11.1.3)

Response

From previous work done on asymmetric cavity pressure hot leg or cold leg breaks, it has been seen that doubling the break area from 1A to 2A does not double loads on components. Lateral forces, moments, and uplift forces on the RV do not follow a linear behavior with respect to break opening area. Loads actually will be consistently lower than the linear ratio. The same relationship is valid for 0.3A and 0.6A breaks. See Table 3.33-1, i.e., multiplying 1A break loads by 2 to determine 2A break loads is a conservative method. The factor of 1.5 was chosen on the basis of our experience in reviewing ACP data from numerous sources (e.g., subcompartment cavity pressure forces on NSS components for various B&W plants). We noted a large variation in the results even considering the differences in cavity configurations and break opening area. However, it was our engineering judgement that 1.5 was a reasonable factor by which LOCA cavity pressure forces should be multiplied. Figure 3.33-1 illustrates the Midland 1 and 2 compartment, which is comparable to Figures 11.1-5 through 11.1-20 of reference 2. Given the same break opening areas and cavities, which are not significantly different from Midland, the factor of 1.5 should adequately account for these differences.

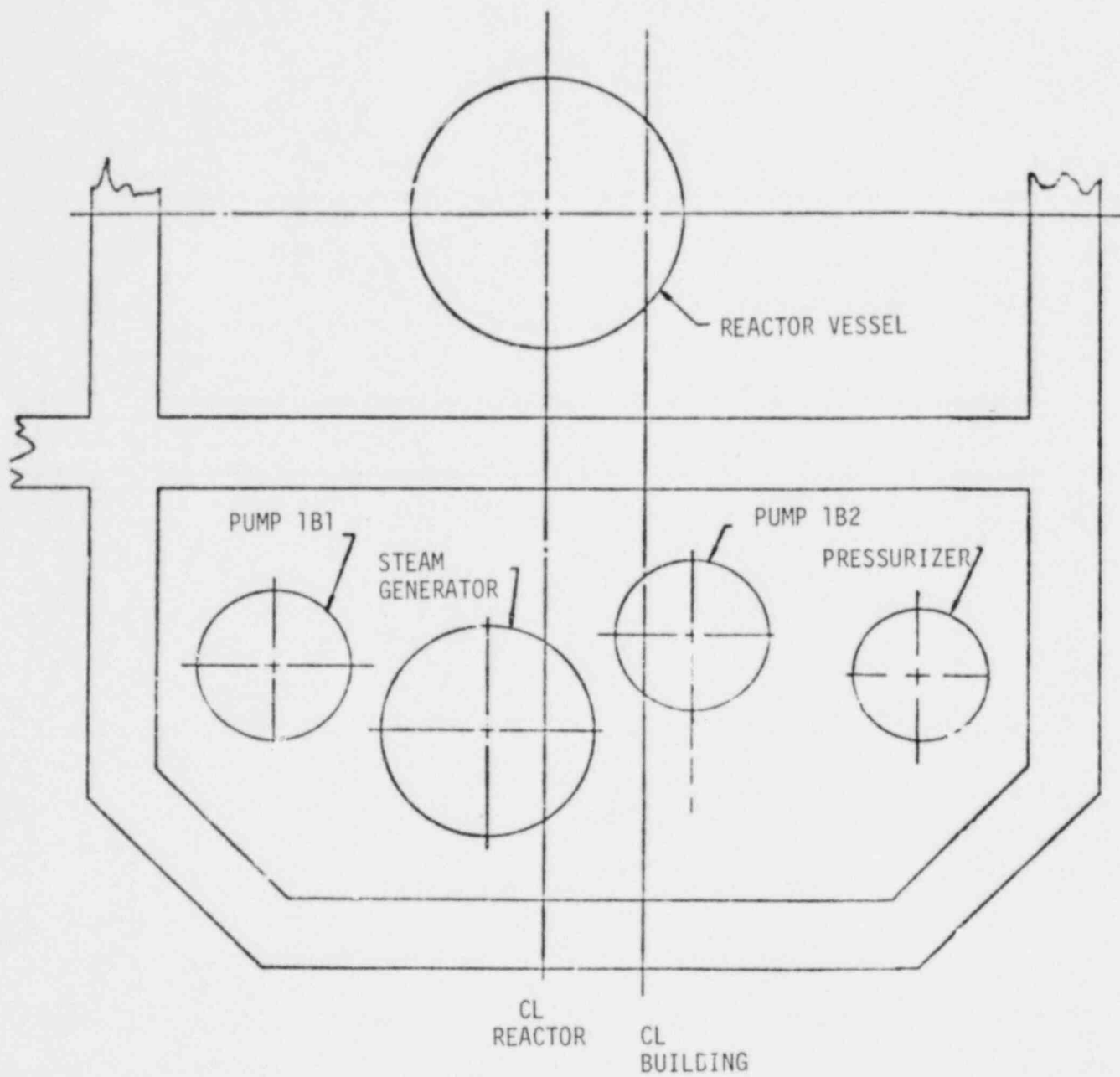
Table 3.33-1. Peak Loads for ANO-1 Reactor Vessel Cavity

		Lateral force, <u>10<sup>6</sup> lb</u>	Moment at skirt, <u>10<sup>6</sup> in.-lb</u>	Uplift force, <u>10<sup>6</sup> lb</u>
Hot leg break	2A	8.94	1973	6.31
	1A	5.41	1178	4.14
	0.6A	3.46	744	2.89
	0.3A	2.10	442	1.75
Cold leg break	2A	3.25	359	3.90
	1A	2.06	260	3.17
	0.6A	1.56	188	2.46
	0.3A	1.04	122	1.34

Notes:

1. For all cases  $2 \cdot (1A \text{ data}) > 2A \text{ data}$ ,  $2 \cdot (0.3A \text{ data}) > 0.6A \text{ data}$ .
2. The ANO data tabulated above are representative of similar data for other plants.

Figure 3.33-1. Midland Unit 2, Loop B -  
Plan View at EL. 634 ft  
(Ref. Dwg C-676, Rev 0)



Note: Unit 1 is opposite hand.

3.34. Clarify Figures 11.1-9 through 11.1-14 with regard to their generation and application. (Section 11.1.3)

Response

Figures 11.1-9 through 11.1-14 of references 2 were part of the data supplied to B&W by Bechtel from their steam generator compartment ACP analysis for Midland 1 and 2 and Davis Besse 2 and 3. The figures are a sample of the ACP forces applied to the RCS model in determining the structural reaction forces.

As an example of their application, the forces from Figure 11.1-9 for a break at the pump discharge were applied as indicated in response to question 3.39 to a mass on the RC pump. The force from Figure 11.1-10 was simultaneously applied in the vertical direction. The forces from Figure 11.1-11 were likewise applied as indicated in response to question 3.39 on the OTSG. In addition, piping reaction forces and jet impingement forces were applied at appropriate locations as described in response to question 3.39.

The data presented in Figures 11.1-9 through 11.1-14 are the basic data, which were then multiplied by the factors described in section 11.1.3 before application to a component.<sup>2</sup> The entire time history was multiplied, time-for-time, not just the peaks.



3.35. Justify that removing the gapped restraints from the model is conservative. (Section 11.1.4.1)

Response

Performing detailed pipe whip calculations that take credit for the effect of gapped restraints in limiting the BOA can only result in BOAs less than or equal to the 2A breaks that were assumed. If the BOAs are reduced as might be expected by restraints designed to limit pipe whip, the ACP forces will be reduced and so will conservatism.

Regarding the dynamic response of the structure, the gapped pipe whip restraints are assumed to be effective only in the piping runs in which a break is postulated. Thus, if a gapped pump restraint were active for a hot leg break, the model would not represent the structure accurately. However, additional restraints would have the general effect of shifting structural modes to higher frequencies and providing additional load paths for the distribution of applied forces.

There is a possibility that a structural frequency might move into resonance with a forcing function, resulting in higher response loads, but that is not likely to increase the design loads significantly. The redistribution of loads to the added load paths could only reduce loads on the major component supports, which we were attempting to assess for structural adequacy.

3.36. Hydraulic compartment pressure and jet impingement are mentioned as loading conditions for the RCS structural analysis. Were dead weight and thermal loadings also considered in the analysis? (Section 11.1.4.1)

Response

Deadweight loads were considered in the simplified pipe stress analysis as stated in section 11.4 of reference 2. Thermal loads were not included because they do not produce primary stresses.

3.37. Expand the discussion concerned with factoring the forces at the break plane. (Section 11.1.4.1)

Response

Suppose a pump suction break as shown in Figure 3.37-1 produces a thrust force,  $T$ , which causes moments at point A that are 1.5 times greater than the pipe can actually transmit without forming a plastic hinge. It is unrealistic to evaluate stresses in the OTSG supports resulting from this applied moment, so  $T$  is multiplied by a factor less than 1.0 in a second iteration of this load case. Because the dynamics of the pipe are important and the model representation is linear elastic, the thrust  $T$  is reduced until subsequent iterations result in the peak moment at point A being within 10% of the elastic limit of the pipe.

3.38. Was the broken pipe loop considered for determining steam generator support loadings? (Section 11.1.4.3)

Response

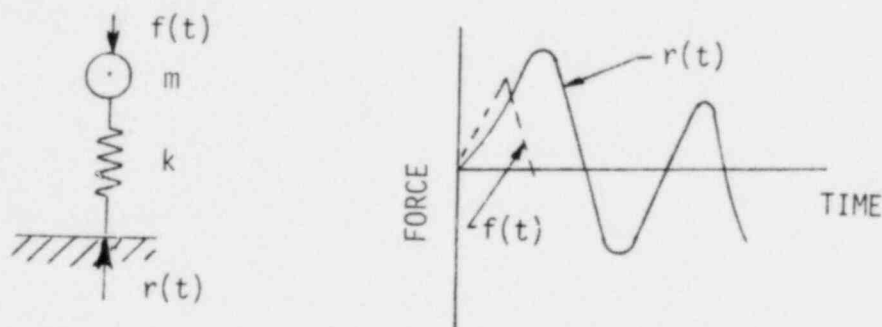
Yes, the broken pipe loop was considered for determining steam generator OTSG support loadings. It is confusing to use the word "loop" to describe portions of the B&W NSS because there are two reactor coolant (RC) pumps associated with each OTSG. "Side" is a less confusing term. Referring to figures in Appendix F of reference 2, it may be seen that only the A side of the RCS has been modeled. The A side consists of the A OTSG and the A1 and A2 cold leg piping. The A side may thus be further subdivided to refer to the A1 and A2 piping.

In determining the OTSG support loads, the entire A side model was active, so whether the question refers to a postulated break in the A hot leg or in the A2 piping, the broken piping was considered. If the break is located so that a plastic hinge would develop in the piping between the break and the next nozzle, then the moments transmitted through that nozzle are equal to the plastic limit, and the adjacent component does experience realistic loading.

- 3.39. Identify in detail what constitutes the applied forces and what constitutes the response loads. Discuss in detail for a specific plant the mathematical model that includes applied force locations and response locations. (Section 11.0)

Response

The figure below illustrates the distinction between applied forces and response loads. The mass,  $m$ , in the single-degree-of-freedom oscillator has an applied force,  $f(t)$ , which is a function of time as shown. The response load at the supporting structure is a function of the structural properties  $m$  and  $k$  (stiffness), and of the nature of the applied force.



In the loading analysis, the applied forces are piping reaction forces\*, ACP forces, and jet impingement forces (if any). For the P1A2 pump discharge break, Table 3.39-1 summarizes the forces applied to the model illustrated in Figures F-1 through F-3 of reference 2.

An example of a source for response loads is element 61, structural joint 23 for the OTSG base (see Figure 11.1-31,<sup>2</sup> a different LOCA case from the example above). Other locations in Table 11.1-3 are similarly chosen. (Note that  $F_y$  for points 25 and 23 in Table 11.1-3 is out of date. The value used in the section 11.2 evaluation of the skirt was 8100 kips compression. This value agrees with Figure 11.1-31.)

-----  
 \*Piping reaction force means reaction to the pressure transient within the RC pressure boundary. It is an applied force and not a response load.

Table 3.39-1. Summary of Applied Forces

<u>Force type</u>	<u>Joint</u>	<u>Direction of application</u>	<u>Comments</u>
Piping reaction	185	X, Y, Z	Summary of multiple fluid from force paths.
	162	Y, Z	See Figure 11.1-6 <sup>2</sup>
	24	Y	
	33	Y	
	13	Y	
	25	X, Y, Z	Summary of multiple fluid force paths.
	44,8	X, Y, Z	Some forces on P1A2 upper cold leg omitted because of plastic hinge consideration.
	55, 154	X, Y, Z	
	37	X, Z	
	64	X, Y, Z	
	93	X, Z	
	168	X, Z	
	188	X, Z	
	ACP	154, 55	X, Y, Z
139, 71		X, Z	
86, 124		X, Z	See Figure 11.1-11 <sup>2</sup>
58, 160		X, Z	
12		X, Z	
2		X, Z	
40		X, Z	
16		X, Z	
Jet impingement	None	--	Jet impingement targets are in the broken piping run and do not affect intact supports.

3.40. Supply results data and comparisons. (Sections 11.2.1 and 11.2.2)

Response

The requested information is included in calculational packages 3.40-1 and 3.40-2.





2) LOWER SUPPORT SKIRT ASSEMBLY

THE PROVIDED ACP LOADS WILL BE COMPARED TO THE LOWER SUPPORT LOADS SHOWN IN THE STRESS REPORT AND CALCULATIONS WILL BE MADE WHEN NECESSARY TO QUANTIFY SIGNIFICANT DIFFERENCES.

A) LOW-LOOP PLANT

ACP LOADS** + DEADWEIGHT Δ	F <sub>x</sub> (KIPS)	F <sub>y</sub> (KIPS)	F <sub>z</sub> (KIPS)	M <sub>x</sub> (FT-K)	M <sub>y</sub> (FT-K)	M <sub>z</sub> (FT-K)
SKIRT @ BASE	2560	8205	5550	42750	2210	190

THESE VALUES WILL BE RESOLVED INTO A RESULTANT HORIZONTAL FORCE (F<sub>R</sub>), A VERTICAL FORCE (F<sub>y</sub>), A RESULTANT OVERTURNING MOMENT (M<sub>R</sub>), AND A TORSIONAL MOMENT (M<sub>y</sub>) FOR DIRECT COMPARISON WITH THE VALUES SHOWN IN MOST OF THE STRESS REPORTS, \*\*REF. 1 Δ REF. 4C

OWNER 177 FA OWNERS GROUP	PROJ NO	CONT NO
SUBJECT PRELIMINARY EVALUATION OF ACP LOADS ON OTSG SUPPORTS	DWG NO	FILE NO
	COMP NO	GROUP NO

GENERAL CALCULATIONS

86-1125142-00

ACP LOADS ** + DEADWEIGHT	FR (KIPS)	F <sub>y</sub> (KIPS)	M <sub>r</sub> ? (FT-K)	M <sub>y</sub> (FT-K)
SKIRT @ BASE	6112	8205	42,751	2210

\*\* REF. 1

THE COMPARABLE ITEMS FROM THE RESPECTIVE STRESS REPORTS FOR EACH CUSTOMER ARE SHOWN BELOW.

STRESS REPORT LOADS	FR (KIPS)	F <sub>y</sub> (KIPS)	M <sub>r</sub> (FT-K)	M <sub>y</sub> (FT-K)	SEE REF. #
DUKE	5220	1825	17,500	0	4a
FLORIDA	5220	1825	17,500	0	4b
TMI-I	5580	2046	25,500	0	4c
TMI-II	4355	1513	26,397	12,470	4d
ARKANSAS	6404	2155	33863	11641	4e
SMUD	8439	1845	90,833	0	4f

IT CAN BE SEEN FROM THE LOADS FOR THE VARIOUS CONTRACTS THAT THOSE IN THE SMUD ANALYSIS ARE THE MOST SEVERE OVERALL.\*

THEREFORE, THE SMUD LOADS WILL BE COMPARED TO THE ACP LOADS.

\* IT IS ACKNOWLEDGED HERE THAT THE SKIRT ASSEMBLY MATERIALS AND GEOMETRIES FOR THE VARIOUS CONTRACTS ARE ESSENTIALLY THE SAME.

CUSTOMER 177 FH OWNERS GROUP	PROP NO	CONT NO
SUBJECT PRELIMINARY EVALUATION OF ACP LOADS ON OTSG SUPPORTS 3.40-4	DWG NO	FILE NO
DATE 1-28-80 DRAWN BY RB DATE 2-14-80	COMP NO	GROUP NO

A REVIEW OF THE CALCULATIONS IN THE SMUD STRESS REPORT (REF. 4A) SHOWS THAT THE LARGE MAJORITY OF EACH SIGNIFICANT STRESS IS GENERATED BY  $F_R$  OR  $M_R$ . THESE  $F_R$  AND  $M_R$  VALUES ON SMUD ARE MUCH LARGER THAN THE ACP  $F_R$  AND  $M_R$  VALUES. THEREFORE, THE ACP LOADS WOULD RESULT IN LOWER STRESSES.

IT IS RECOGNIZED THAT THE ACP VALUES OF  $F_y$  AND  $M_y$  ARE SOMEWHAT LARGER THAN THE SMUD  $F_y$  AND  $M_y$ . HOWEVER, AS WAS IMPLIED ABOVE, THESE ARE ONLY MINOR CONTRIBUTORS TO THE TOTAL SIGNIFICANT STRESSES (I.E. -  $F_y$  GIVES FLANGE BEARING CONTRIBUTION  $\approx 0.8$  KSI;  $M_y$  GIVES A SHEAR CONTRIBUTION OF  $\approx 0.6$  KSI, BOTH OF WHICH ARE INSIGNIFICANT).

CONCLUSION:

IT IS CONCLUDED THEN THAT SINCE THE SMUD STRESS REPORT RESULTS SHOW THE SUPPORT SKIRT ASSEMBLIES OF ALL THE LOW-LOOP PLANTS TO BE ACCEPTABLE FOR LOADS HIGHER THAN THE ACP LOADS, THE ACP LOADS ARE ACCEPTABLE.

OWNER 177 FA OWNERS GROUP	PROP NO	
SUBJECT PRELIMINARY EVALUATION OF ACP	DWG NO	FILE NO
LOADS ON OTSG SUPPORTS 3.40-5	COMP NO	GROUP NO
LEW DATE 1-28-80 DRAWN BY RB DATE 2-14-80		A/5

REFERENCES:

- 1) BFW DOCUMENT # 86-1106948-02 (ACP LOADS)
- 2) FOR DUKE - BFW DWG. 112793 D-0  
 FOR TMI-I - BFW DWG. 112794 D-10/20/71  
 FOR TMI-II - BFW DWG. 112795 D-3  
 FOR FLORIDA - BFW DWG. 112796 D-1  
 FOR ARKANSAS - BFW DWG. 112797 D-2  
 FOR SNUD - BFW DWG. 112799 D-3
- 3) BFW SPECIFICATION 3002/NSS-14/1077  
 FOR REACTOR COOLANT SYSTEM FOUNDATION AND  
 NOZZLE LOADINGS
- 4) STRESS REPORT FOR STEAM GENERATOR, REPORT  
 #12, "STRESS ANALYSIS OF STEAM GENERATOR  
 SUPPORT SKIRT" -
  - a) BFW CONTRACT 620-0003/4/9-55
  - b) BFW CONTRACT 620-0007-55
  - c) BFW CONTRACT 620-0005-55
  - d) BFW CONTRACT 620-0006-55
  - e) BFW CONTRACT 620-0008-55
  - f) BFW CONTRACT 620-0011-55
- 5) STRESS REPORT FOR STEAM GENERATOR, REPORT  
 #12, "STRESS ANALYSIS OF STEAM GENERATOR  
 SUPPORT SKIRT AND ATTACHMENTS FOR MECH  
 ANICAL LOADS", BFW CONTRACT 620-0014-55

CUSTOMER 177 FH OWNERS GROUP	PROP NO	CONT NO
SUBJECT PRELIMINARY EVALUATION OF ACP	DWG NO	FILE NO
LOADS ON OTSG SUPPORTS 3.40-6	COMP NO	GROUP NO
LEW 1-28-80 R3	2-14-80	3/3

CALCULATION DATA/TRANSMITTAL SHEET

DOCUMENT IDENTIFIER      CALC. 32 - 1125237      - 00

TRANS. 86      -      -

TYPE:       RESEARCH & DEVELOPMENT       SAFETY ANALYSIS REPORT       NUC. SERV. INPUT       DESIGN RQMT.       DESIGN VERIF.       OTHER

TITLE OTSG SUPPORT SKIRT FILLET WELD STRESS ANALYSIS - ACP PHASE II

PREPARED BY LARRY E. WHITE      REVIEWED BY D.E. Thoren

TITLE ENGINEER      DATE 4-3-81      TITLE Principal Engr.      DATE 4/21/81

PURPOSE:      THE PURPOSE OF THIS CALCULATIONAL PACKAGE IS TO PERFORM THE STRESS ANALYSIS OF THE 1" FILLET WELD (WG-194) THAT JOINS THE OTSG SUPPORT SKIRT GUSSET TO THE TOP + BOTTOM CENTER SUPPORT DISKS.

SUMMARY OF RESULTS (INCLUDE DOC. ID'S OF PREVIOUS TRANSMITTALS & SOURCE CALCULATIONAL PACKAGES FOR THIS TRANSMITTAL)

THE AVERAGE SHEAR STRESS ON THE THROAT OF THE 1" FILLET WELD IS CALCULATED TO BE 25,288 PSI WHICH IS LESS THAN THE SHEAR ALLOWABLE OF 35,280 PSI.

THE PRIMARY SOURCE CALCULATIONAL PACKAGE IS B+W DOCUMENT # 32-1119695-00, "OTSG SUPPORT - WELD STRESS ANALYSIS, PHASE II", BY D.E. THOREN, DATED 6-13-80.

DISTRIBUTION

DISCUSSION:

THE CALCULATIONS CONTAINED HEREIN ARE PERFORMED IN ORDER TO DEMONSTRATE THE STRUCTURAL ADEQUACY OF THE 1" FILLET WELD (TOP/BOTTOM SUPPORT DISC-TO-SUPPORT GUSSET WELD) FOR ASYMMETRIC CAVITY PRESSURE LOADS - PHASE II. THE SUBJECT WELD IS DESIGNATED AS WG-194 ON REFERENCE 1.

THE SIGNIFICANT LOAD ON THIS WELD RESULTS FROM THE DOWNWARD COMBINED DEADWEIGHT AND LOCA FORCE OF 10,083 KIPS (PER REF. 1, SHT. 12).

A FINITE ELEMENT ANALYSIS OF THIS WELD HAS DETERMINED THE FORCES ACTING ON THE WELD DUE TO THIS LOAD (REF. 1, PG.'S 19 + 20).

AS WAS DONE IN THE STRESS REPORT (REF. 2), THE AVERAGE SHEAR STRESS ON THE THROAT AREA OF THE WELD WILL BE CALCULATED AND COMPARED TO THE ALLOWABLE STRESS.

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WELD SHEAR STRESS ALLOWABLE:

PER REF. 4 TABLE F-1322.2-1, ELASTIC SYSTEM, COMPONENT SUPPORTS, STRESS LIMITS, THE LIMIT FOR THE BASE METAL IS:

1.5 S<sub>m</sub> OR 1.25 S<sub>y</sub>, WHICHEVER IS GREATER BUT NOT TO EXCEED 0.75 S<sub>u</sub>

THE MATERIALS INVOLVED ARE:

TOP + BOTTOM SUPP. DISC	SA-302 GR B	} REF. <u>6</u>
CENTER SUPP. FORGING	SA-541 CL 2	
SUPP. GUSSET	SA-302 GR B	

THESE ARE ALL MAG MOLT MATERIALS WITH THE FOLLOWING PROPERTIES FROM REF. 4:

S<sub>y</sub> = 47,100 PSI\*      S<sub>m</sub> = 26,700 PSI\*      S<sub>u</sub> = 80,000 PSI\*

1.25 S<sub>y</sub> = 56,520 PSI  
 1.5 S<sub>m</sub> = 40,050 PSI  
 0.75 S<sub>u</sub> = 56,000 PSI

∴ 56,000 PSI IS ALLOWABLE; HOWEVER, TAKING INTO ACCOUNT THE 5% INCREASE ON ULTIMATE STRENGTH PERMITTED BY REF. 5, THE TOTAL ALLOWABLE BECOMES 1.05 (56,000) = 58,800 PSI

\* PER REF. 3, THE WELD TEMPERATURE WILL BE LESS THAN 200°F, THUS, PROPERTIES ARE TAKEN @ 200°F

THUS, THE SHEAR ALLOWABLE (APPLICABLE TO THE WELD)  
IS:

$$T_{\text{allow.}} = (.6 \times 58,800) = \underline{\underline{35,280 \text{ PSI}}}$$

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FORCES IN THE WELD:

1) AVERAGE FORCE IN WELD PARALLEL TO GUSSET LENGTH -

FROM REF. 1, PG. 19, THE RESULTING FORCES ALONG THE LENGTH OF THE WELD ARE SHOWN. THESE FORCES AVERAGED ALONG THE LENGTH YIELD -

$$2.2 \text{ " (273,160 + 2(87,585) + 2(50,940) + 2(29,920) + 2(16,560) + 1070) } \\ \frac{5(2.2) (2)}{\text{SEGMENT LENGTH}}$$

$$= P_{\text{AVE}} = \frac{2.2}{(2)11.0} [653,480] = 65,348 \text{ \#/SEGMENT}$$

2) AVERAGE FORCE IN WELD IN UPWARD DIRECTION -

FROM REF. 1, PG. 20, THE RESULTING FORCES IN THE UPWARD DIRECTION ARE SHOWN. THESE FORCES AVERAGED ALONG THE LENGTH OF THE WELD YIELD -

$$2.2 \text{ (180,050 + 2(89,790) + 2(58,250) + 2(40,000) + 2(26,560) + 9670) } \\ \frac{5(2.2) (2)}{\text{SEGMENT LENGTH}}$$

$$= P_{\text{AVE}} = \frac{2.2}{(2)11} [619,390] = 61,934 \text{ \#/SEGMENT}$$

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## 3) RESULTANT SHEAR LOAD ON WELD THROAT:

LOAD PER INCH OF WELD -

$$P_{X \text{ AVE}} / 4.4" = 65,348 / 4.4 = \underline{14,852 \# / \text{IN}}$$

← LENGTH OF WELD PER SEGMENT CONSIDERING  
WELD ON BOTH SIDES OF GUSSET THICKNESS

$$P_{Y \text{ AVE}} / 4.4" = 61,934\# / 4.4" = \underline{14,076 \# / \text{IN}}$$

RESULTANT - YEAR FORCE:

$$Q = \left[ (14,852)^2 + (14,076)^2 \right]^{1/2} = \underline{20,462 \# / \text{IN}}$$

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WELD STRESS :

$$\tau = Q/A = 20,462 / (0.707)(1.0")$$
$$\underline{\underline{28,939 \text{ PSI}}}$$

THIS IS LESS THAN THE ALLOWABLE SHEAR STRESS OF 35,280 PSI (FROM SHEET 4); THEREFORE, THE 1" FILLET WELD (WG-194) IS ACCEPTABLE FOR THE PHASE II ACP LOADS.

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REFERENCES

- 1) B+W CALCULATIONAL DOCUMENT # 32-1119695-00  
"OTSQ SUPPORT - WELD STRESS ANALYSIS, PHASE II",  
BY D. E. THOREN, DATED 6-13-80.
- 2) STRESS REPORT FOR STEAM GENERATOR, REPORT #12,  
"STRESS ANALYSIS OF STEAM GENERATOR SUPPORT  
SKIRT & ATTACHMENTS FOR MECHANICAL LOADS", B+W  
CONTRACT 620-0014-55.
- 3) STRESS REPORT FOR STEAM GENERATOR, REPORT # 3  
"STRESS ANALYSIS OF SUPPORT SKIRT", B+W CONTRACT  
620-0014-55, PAGE B-6-5.
- 4) ASME CODE, SECTION III, 1977 EDITION, APPEND-  
ICES.
- 5) ASSUMPTIONS, METHODOLOGY, AND ACCEPTANCE CRITER-  
IA REPORT FOR ASYMMETRIC LOCA LOADS EVALUATION,  
PHASE II, BAW-1538, REV. 1, APRIL 1979.
- 6) "LIST OF MATERIAL, STEAM GENERATOR, SHEET #2",  
B+W DRAWING # 151925 E, REV. 8.
- 7) "ASSEMBLY AND DETAIL OF SUPPORT SKIRT", B+W  
DRAWING # 151930 E, REV. 9.

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3.41. Supply justification that all of the 177-FA skirt-supported plants are hydrodynamically equivalent. Also include a list of differences, if any. (Section 2, Appendix C)

Response

The 177-FA skirt-supported plants have internals that are dimensionally identical except for manufacturing tolerances. Historically, the 177-FA skirt-supported plants have been grouped into a single generic category as shown in BAW-10103, Rev 1<sup>4</sup>, sections 1.2 and 3; and BAW-10104, Rev 1<sup>5</sup>, section 4.1 and Table 1-1. The responses to questions 2.10, 2.18, and 3.32 provide additional qualifications that these plants are hydrodynamically equivalent.

- 3.42. Is the hydrodynamic mass matrix changed as the annulus changes size during the dynamic analysis? What experimental/analytical correlations exist for using this technique for this type of analysis? Have benchmark cases been completed? Enclose the results of these analyses. (Section 3.3, Appendix D)

Response

The hydrodynamic mass matrix was assumed to be constant as the annulus changes size. The justification to this assumption is that by design, the reactor core will not deflect more than 1/4 inch during a LOCA, in an annular gap width of about 10 inches. The change in nodal volume is therefore negligible, resulting in a hydrodynamic mass matrix that is more or less constant as the core moves.

B&W has conducted extensive analytical and experimental studies to verify the hydrodynamic mass approach to steady-state vibration analysis of coupled fluid-shell systems. The results of these benchmark studies were presented to the NRC and EG&G on March 14, 1979. B. F. Saffell and R. W. Macek from EG&G represented the NRC as consultants at this presentation. Three NRC staff members were also present for all parts of this presentation.

Reference 29, sections 3, 4, 5, and 6 and Appendix A discuss benchmark studies, including experimental tests.

3.43. Describe in detail how the core bounce results are used in the system dynamic analysis. (Section 5.2.2)

#### Response

The core bounce analysis is a vertical dynamic analysis of the internals for LOCA. The model includes a vertical representation of the core support cylinder, lower grid, fuel assemblies, upper grid, plenum cylinder, column weldments, and plenum cover. The internals are supported vertically from the reactor vessel ledge via the plenum cover, and the core support cylinder and fixed boundary conditions at the RV ledge terminate the core bounce model. This analysis includes the linear internals along with the nonlinear effects of the fuel assemblies and their movement between the upper and lower grids. For the core bounce analysis the internal pressures during the LOCA transient are integrated over the vertical area of each part of the internals, producing force time histories. These time histories are then applied to the structures and the response recorded. The total resulting force at the reactor vessel ledge is applied to the reactor vessel isolated model.

For the reactor vessel isolated model, the internal pressures are integrated over the vertical area of the reactor vessel to create a force time history. This time history is applied to the reactor vessel in the vertical direction along with core bounce time history at the reactor vessel flange. Thus, the reactor vessel sees the total integrated vertical pressure force from the internals and the force caused by the differential pressure in the upper and lower vessel heads integrated over the vertical surface area of the heads along with the dynamic response of the internals.

The vertical displacement response of the reactor vessel in the isolated model feeds back into the internals (linear components). This feedback loop excites the mass of the internals in the vertical direction causing a time history response. This time history response (caused by vertical vessel displacement) can be time-phased with the linear structural response in core bounce.

The method of analysis described above allows necessary detail in the nonlinear core bounce while permitting the analysis to be performed in reasonable computer time. If the linear and the nonlinear are combined, the computer time is extensive, and study of separate effects is questionable.

3.44. Supply the following (Appendix E):

- a. Boundary conditions used in the analyses.
- b. Expanded mathematical model description.
- c. Detailed discussion of the applied loads.

Response

The finite element skirt model included the lower end of the reactor vessel and was built to determine stiffness values for the skirt and to evaluate the load-carrying capability of the support.

In the stiffness study the base of the skirt was fixed by fixing the flange base at each bolt location for the tension side of the skirt. At the same time the compression side of the skirt was fixed at the base of the flange. The flange was also fixed horizontally to shear.

A force was applied, and the resultant deflection and rotation were used to calculate the flexibility terms. A moment was then applied, and again flexibility terms were calculated. These terms were then combined to form the skirt flexibility matrix, which was used to determine equivalent beam section properties to represent the skirt in the linear model.

The loading capability evaluations were performed using ANO-1 foundation spring rates. The individual bolt stiffness was modeled at the bolt locations, and these became active when their preload was overcome. Prior to this point the foundation stiffness (local concrete) was active for the tensile and compressive loadings. These were distributed evenly at the base of the flange. The shear stiffness of the foundation was input radially to the flange.

The model, including the boundary conditions, represented the flange and skirt geometry including the holes in the support skirt. It should be noted that the minimum skirt thickness of 2 inches was used so that the load carrying capability as well as the skirt flexibility were conservatively low and high, respectively.

The only results actually used in the analysis were the flexibilities, which were used in the beam model. The load-carrying capability evaluations confirmed that ASME Code allowables were reasonable; therefore, the code allowables were used in the final stress analysis of the skirts.



3.45. Specify and justify the value of  $f'_c$  used in the analysis. (Appendix I)

Response

The compressive strength  $f'_c$  of concrete used for the cavity wall analysis is given in Table 3.45-1. For the linear analysis, the 28-day minimum specified strength was used and is conservative. For the nonlinear analysis of the TMI-1 cavity wall, the 90-day strength was used as a more realistic value of  $f'_c$  (see section 6.2.14 of reference 2 for further discussions).

Table 3.45-1. Concrete Compressive Strength  
for Cavity Walls

<u>Plant</u>	<u>Compressive strength <math>f'_c</math>, psi</u>	<u>Reference</u>
Oconee	5000	Duke Power Co. Dwg O-1069A, Rev 7
TMI-1	5000 <sup>a</sup> 6650 <sup>b</sup>	Gilbert Assoc. Dwg D-421-019, Rev 1
TMI-2	5000	Burns & Roe Dwg 4157, Rev 8
Crystal River	5000	Gilbert Assoc. Dwg SC-421-012, Rev 5
ANO-1	5500	Bechtel Corp. Dwg C-167, Rev 6
Rancho Seco	5000	Bechtel Corp. Dwg C-309, Rev 2
Davis Besse	5000	Bechtel Corp. Dwg C-154, Rev 5

<sup>a</sup>Linear analysis.

<sup>b</sup>Nonlinear analysis.

3.46. Discuss in detail all boundary conditions. (Appendix I)

Response

Ocone Cavity Wall (Figure I-2  
of reference 2)

1. The base of the model is fixed in all three directions at the cavity wall/pedestal intersection because this condition gives the most conservative base reactions. For a cylindrical structure, the edge boundary conditions have limited effects on the primary stresses for pressure loads.
2. The edges of the fuel canal slab and fuel canal walls are fixed in all directions at the interface with the secondary shield walls. The effects of the stiffnesses of the secondary shield walls on the primary cavity shield wall are not significant.
3. The core flood tank walls are modeled with equivalent trusses in three directions. The stiffness of these trusses was taken as the lower bound values, so that more loads were carried by the cavity wall.
4. At the line of structural symmetry (through the hot legs), the displacements normal to the plane of the symmetry are fixed for the symmetric load analysis (symmetric boundary conditions), and the displacements parallel to the plane of symmetry are fixed for the asymmetric load analysis (asymmetric boundary condition).

TMI-1 Axisymmetric Cavity Wall Nonlinear  
Analysis Model (Figure I-3 of reference 2)

1. At the pedestal/liner plate interface, only the vertical displacements are restrained. At the outside face of the pedestal base, equivalent springs are modeled to account for the base concrete slab extending outward above the liner plate to the edge of the containment building.
2. At the junction of the fuel canal slab, equivalent springs are used to account for the fuel canal slab and walls. These springs are lower bound values and were calculated from detailed finite element models.

TMI-1 Fuel Canal Slab Model (Figure I-4  
of reference 2)

1. The fuel canal slab and walls are fixed in all three directions at the interface with the secondary shield walls.
2. A symmetric boundary condition is used at the line of symmetry.

TMI-2 Cavity Wall Model (Figure I-5  
of reference 2)

1. The base of the model is fixed in all directions at the cavity wall/pedestal interface.
2. The stiffness of the secondary shield wall is modeled by equivalent truss elements in three directions at the interface with the fuel canal slab and walls.
3. The core flood tank walls are represented by equivalent truss elements in three directions.
4. Appropriate symmetric or asymmetric boundary conditions are used at the line of symmetry.

CR-3 Cavity Wall Model (Figures I-6  
and I-7 of reference 2)

1. The model base is fixed in all directions at the cavity wall/pedestal interface.
2. The fuel canal slab and walls are fixed in all directions at the secondary shield wall interfaces.

ANO-1 Cavity Wall Model (Figure I-8  
of reference 2)

1. The model base is fixed in all three directions at the cavity wall/pedestal interface.
2. There is an elastomer between the fuel canal floor and the top of the cavity wall. The top of the cavity wall was modeled as free to displace in all directions.
3. Appropriate symmetric or asymmetric boundary conditions are used at the line of symmetry.

Rancho Seco Cavity Wall Model (Figure I-9 of reference 2)

1. The model base is fixed in all three directions at the cavity wall/pedestal interface.
2. The edges of the fuel canal slab and walls are fixed at the secondary shield wall interfaces.
3. Appropriate symmetric or asymmetric boundary conditions are used at the line of structural symmetry.

Davis Besse Cavity Wall Model (Figure I-10 of reference 2)

1. The model base is fixed in all three directions at the cavity wall/pedestal interface.
2. The stiffness of the secondary shield wall is modeled with equivalent springs in three directions at the interface with the fuel canal slab.
3. The boundary of the core flood wall is fixed in the strong direction and modeled with equivalent springs in the weak direction of the secondary shield wall.
4. The boundary of the fuel canal wall is fixed in the strong direction and free in the weak direction of the secondary shield wall.
5. Symmetric boundary conditions were used at the model edges for hot leg break analyses.

3.47. Define the applied loadings, including application points. (Appendix I)

Response

The pressure load developed in the cavity was applied to the inner surfaces and penetrations of each cavity wall using the pressure load option of EDS-SNAP. Typical pressure functions and the element surfaces where these functions are applied are shown in Figure 3.47-1. Forty-six pressure functions and 60 pressure element surfaces are used in this example. Detailed descriptions of the actual pressure-time histories used in the analysis are given in Appendix B.

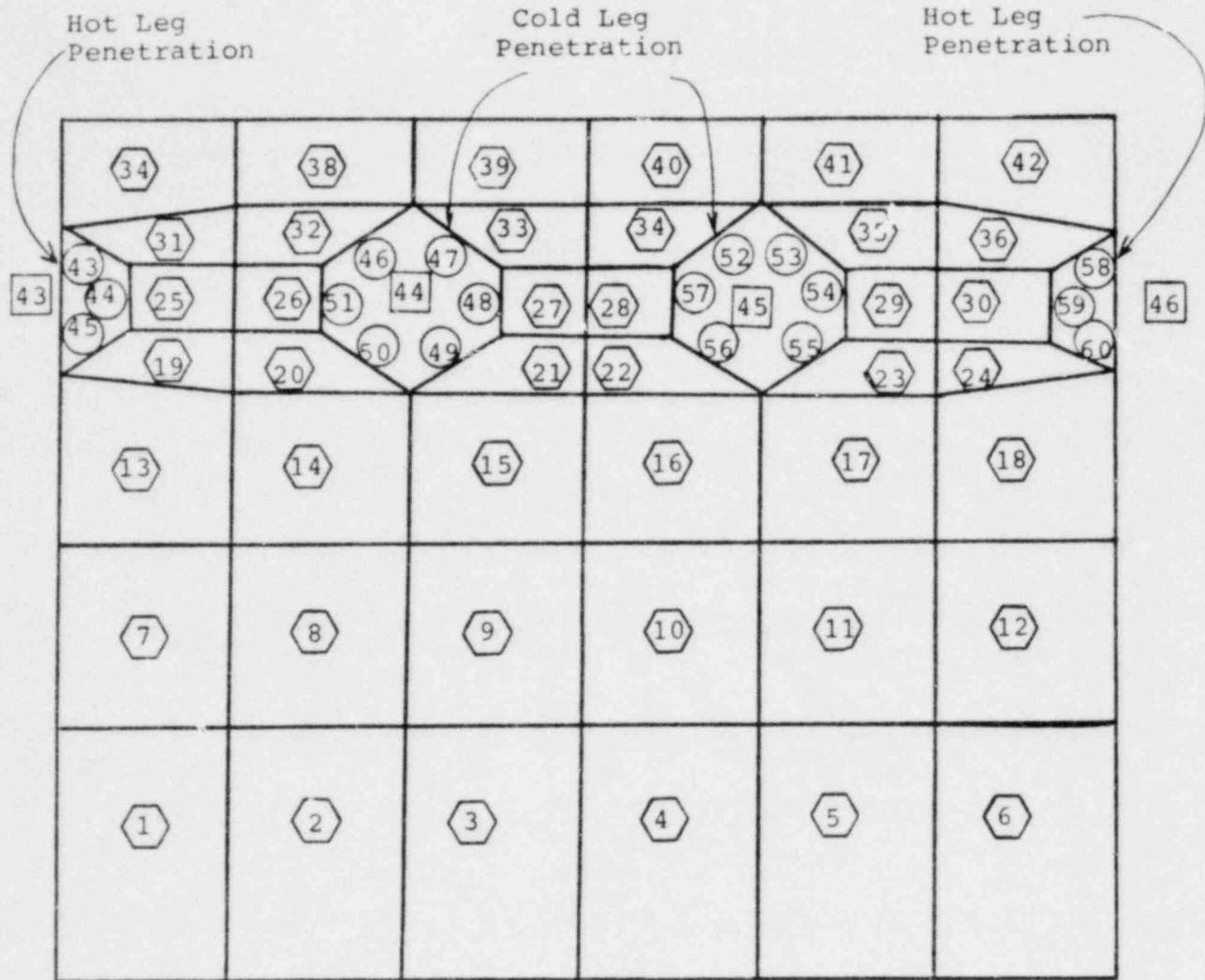
EDS-SNAP element pressures are developed from CRAFT2<sup>1</sup> pressures as follows: the pressures are weighted with respect to the surface areas of the EDS-SNAP elements and CRAFT2 volumes so that the net loads on the cavity walls are maintained. The pressures are taken as constant over the elements since the pressure gradients at a time are small, and also as the EDS-SNAP elements are typically smaller than CRAFT2 elements. Where EDS-SNAP element surfaces are larger than corresponding CRAFT2 volumes, pressures from the CRAFT2 volumes are extended to conservatively apply pressure to the total EDS-SNAP element surface. Figure 3.47-2 shows EDS-SNAP pressure surfaces and the corresponding CRAFT2 pressure volumes.





The data in Appendix B are divided into four sections:

1. Input to program MAP.
2. CRAFT2 volume pressure data.
3. EDS-SNAP element pressure-time histories.
4. Integrated net thrust load on total cavity wall.

For example, in section 3, EDS-SNAP pressure-time history 30 is for the EDS-SNAP element adjacent to the hot leg penetration where a 2.0A break occurs. The time history is given from time 0.0000 to 1.9996 seconds. This pressure-time history is applied over the entire exposed surface area of the EDS-SNAP element.

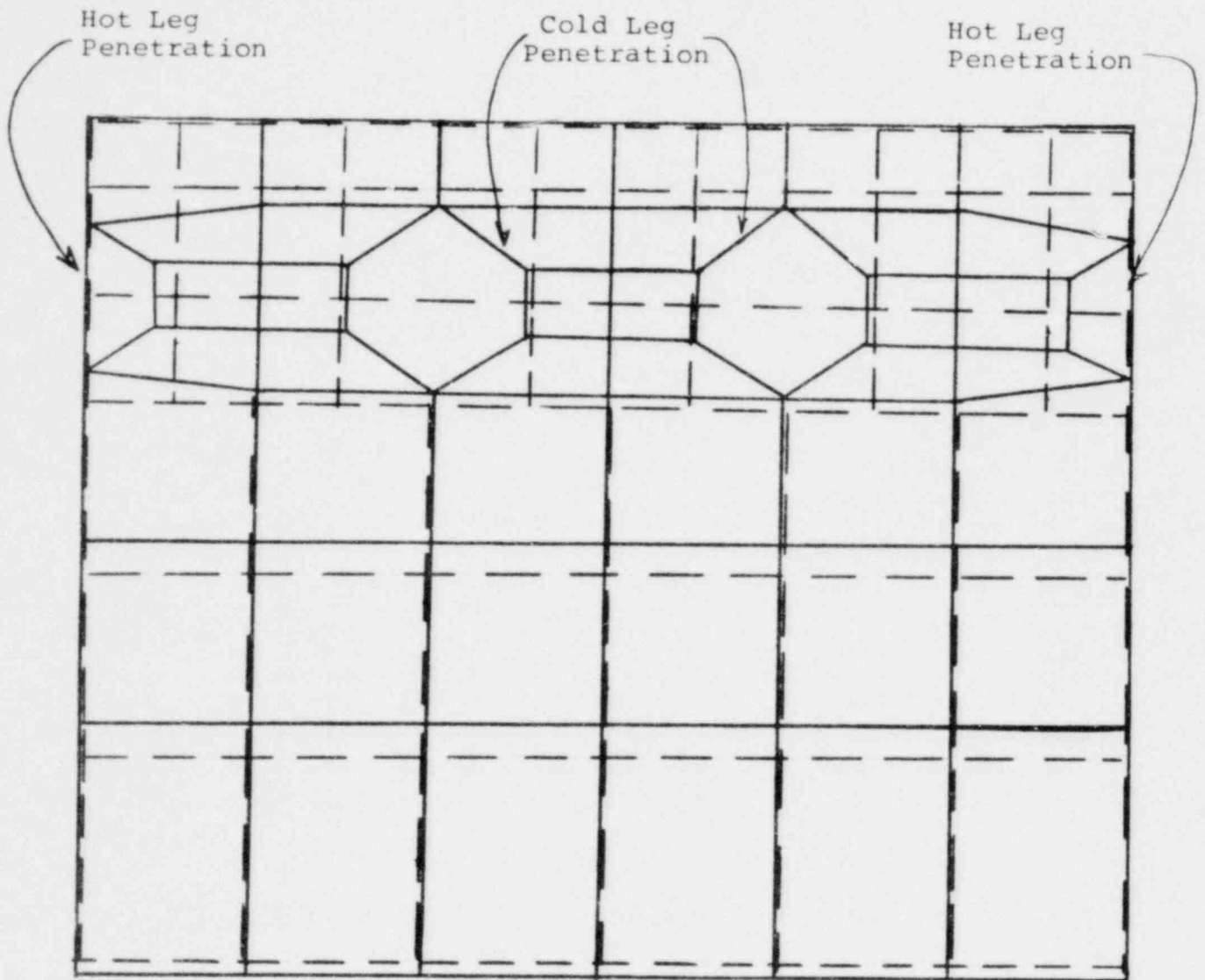
Figure 3.47-1. TMI-2 Pressure Surfaces and Functions



  60 Pressure Surfaces  
  46 Pressure Functions

Note: This figure is an expanded view of the inside surfaces of the 180° TMI-2 cavity wall model.

Figure 3.47-2. TMI-2 Pressure Surfaces and Volumes



Solid Lines - EDS-SNAP Pressure Surfaces  
Dotted Lines - CRAFT2 Pressure Volumes

Note: This figure is an expanded view of the inside surfaces of the 180° TMI-2 cavity wall model.

3.48. Discuss in detail the procedure used for handling asymmetric loadings.  
(Appendix I)

Response

For a structure with structural symmetry, we can decompose any arbitrary loads with symmetric and asymmetric values and analyze only half of the structure.

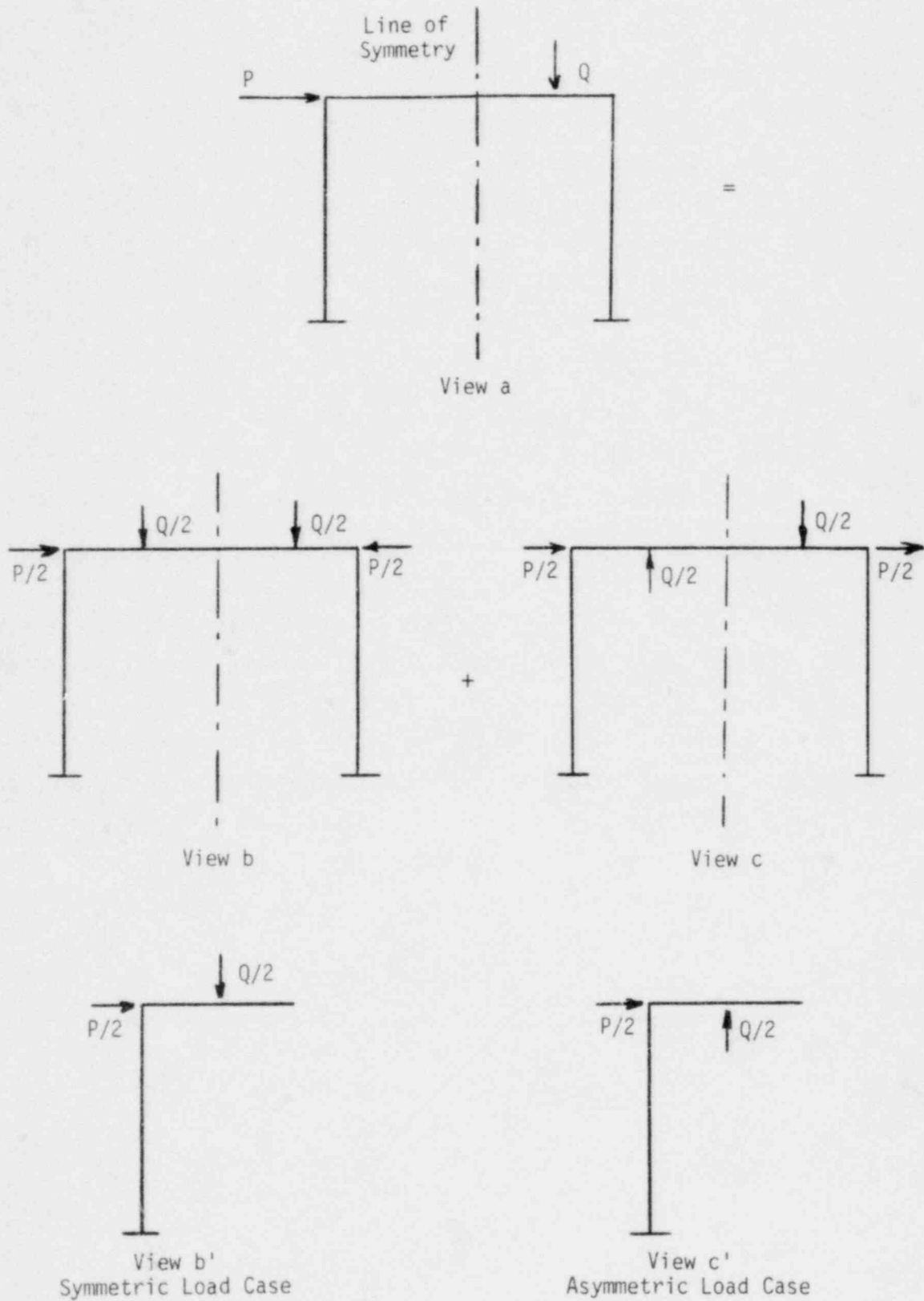
For example, consider a simple, two-dimensional frame as shown in Figure 3.48-1, View a. The structure has a line of symmetry as indicated by the dashed line. Arbitrary loads P and Q are applied to the structure. The loads are divided into symmetric and asymmetric loads as shown in Views b and c. These load cases can then be analyzed with half of the structure as indicated in Views b' and c'.

In applying this procedure to the cavity wall analysis, the pressure data for a full 360° model was first developed. The analyses were then performed with symmetric and asymmetric loads using one half of the cavity wall. The final results were obtained by combining these separate analyses.

Specifically, for the hot leg break case analyses, the structural and pressure loads are both symmetric. Hence, only one symmetric structure analysis was performed. For the cold leg break case analyses, the structure is symmetric about the hot leg centerline, but the loads are not. Thus, two analyses were performed, one with the symmetric portion of the cold leg pressures and one with the asymmetric portion. The results from these two analyses were then combined to give the exact stress state at all locations in the cavity wall.



Figure 3.48-1. Two-Dimensional Frame



3.49. Is using isotropic properties conservative for loads other than in the hoop direction? (Appendix I, Section 2)

Response

When static analyses of the cavity walls were performed, the orthotropic analysis gave lower hoop stresses and higher bending stresses than the isotropic analysis. Thus, using isotropic properties is not conservative for bending loads at the cavity wall/pedestal interface.

The primary load-carrying mechanism of the cavity walls is hoop tension. Using isotropic properties for the analysis predicts upper bound hoop loads. If the hoop steel can take these loads, without yielding, then the cavity wall is able to withstand the LOCA event. Therefore, we did not perform further analyses using a cracked concrete orthotropic material law.

However, if we found some hoop steel yielding from isotropic analysis results, we reran the analysis using an orthotropic material law to model the cracking of concrete. Bending moments at the cavity wall/pedestal interface in the orthotropic analysis were higher than those in the isotropic analysis. These effects are localized to the base of the cavity wall. Bending moments are carried by vertical steel in the cavity wall.

3.50. In light of the fuel canal slab strength reduction, discuss bending and strength at discontinuities and boundaries. (Section 3.1, Appendix I)

Response

The fuel canal slab reduction assumption for the Oconee cavity wall was adopted in the analysis to conservatively predict hoop stress in the wall.

The primary mode of cavity wall resistance to the pressure load is through hoop tension. The fuel canal slab restrains radial expansion and thereby causes lower hoop stresses than might otherwise occur. Thus, by making the fuel canal slab weak in comparison to the pedestal, a larger portion of the load is carried by hoop steel. As the hoop steel carries this load, below yield stress, this modeling assumption results in a conservative load-carrying capacity analysis of the cavity wall.

As can be seen in Tables 10.10-1 and 10.10-2 of reference 2, the shear and bending stress ratios are low: they are not expected to be excessive if the analysis was performed with the fuel canal slab at full strength. The steady-state isotropic analysis of the cavity wall with reduced slab stiffness, however, gave an upper bound prediction of the hoop stresses. This ensures the functional adequacy of the cavity wall even if there should be local yielding at the boundary interfaces due to excessive moments.

3.51. What effect does the water mass have on impact values? (Appendix J, Section 1)

Response

In developing the STARS model of the fuel assembly, the water mass was added to the fuel assembly node masses. This increase in the fuel assembly mass due to water mass tends to reduce the fuel assembly frequencies and increase the fuel assembly impact loads. This is conservative assumption since it yields higher loads on the fuel assembly components.

3.52. Are grid-to-grid-to-baffle and grid-to-baffle impact stiffnesses verified? Supply the details. (Appendix J, Section 1)

Response

Yes. The grid-to-grid-to-baffle and grid-to-baffle impact stiffnesses are verified by testing.

The single spacer grid testing as shown in Figure 3.52-1 shows the test configuration to determine the spacer grid stiffness, elastic impact load, and damping ratio. This test configuration simulates the grid-to-baffle impact because the spacer grid with loaded rods is impacted against a very stiff impact surface.

Double spacer grid testing was also performed to determine the stiffness for a grid-to-grid-to-baffle impact. Both possible configurations were tested as shown in Figures 3.52-2 and 3.52-3 to simulate such impacts. Test results indicated that the spacer grid properties, such as stiffness, elastic impact load limit, and damping ratio, do not change for both these configurations from those obtained from the single spacer grid testing. Therefore, the spacer grid properties from single spacer grid testing are used for both types of impacts.

Figure 3.52-1. Single Grid Impact

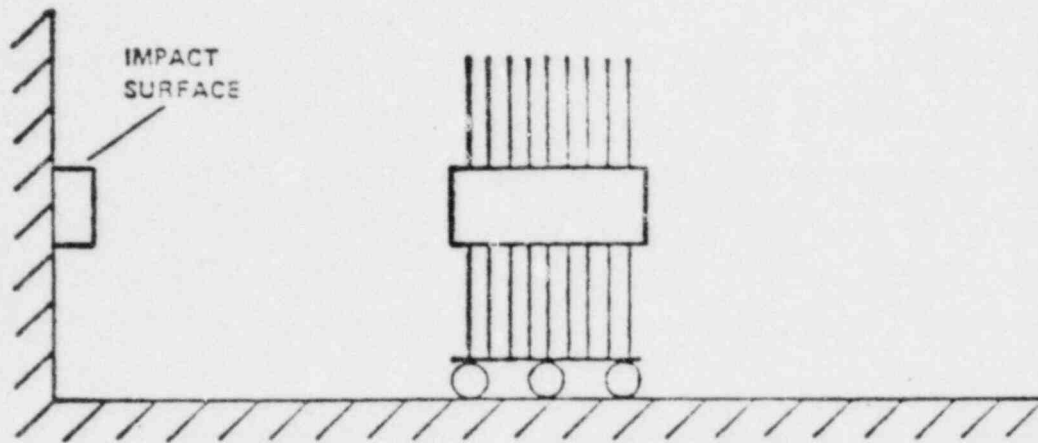


Figure 3.52-2. Double Grid Impact Configuration 1

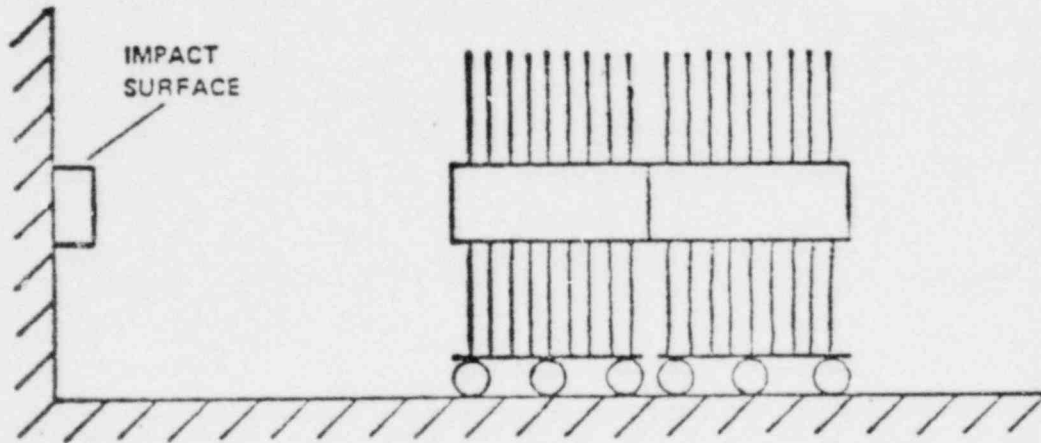
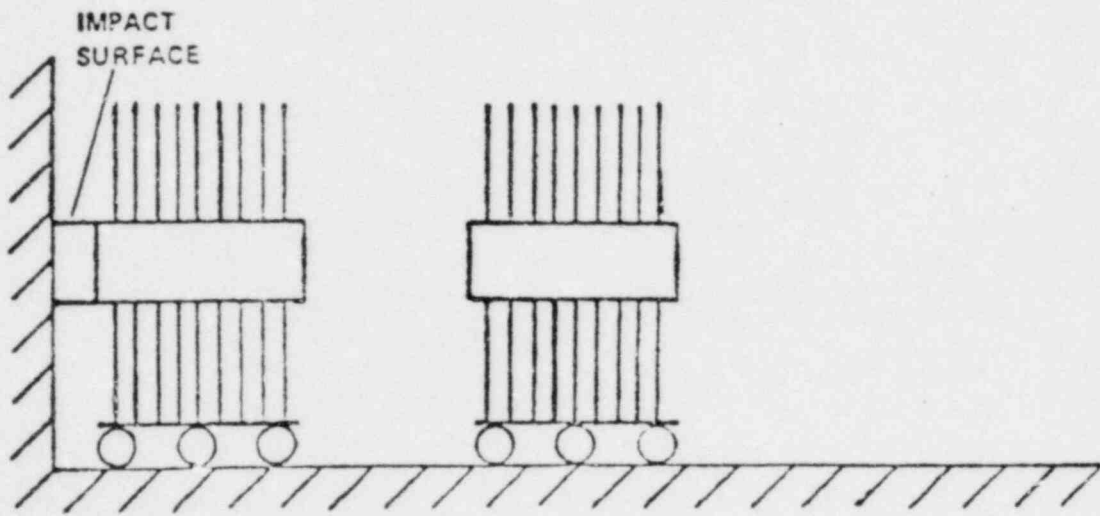


Figure 3.52-3. Double Grid Impact Configuration 2



- 3.53. Supply the following information: a. Detailed math models  
(Appendix K) 1. Nodalization  
2. Applied Force Locations  
b. Model boundary conditions

#### Response

Four embedment models were used. Clarification on the nodalization, applied force locations, and boundary conditions for three of these models is provided in this response. The fourth, the nonlinear embedment model, is of relatively simple configuration and is fully described in section 2.3, Appendix K, of reference 2.

#### 1. Generic Embedment Model

The generic embedment model is shown in Figures 3.53-1 and 3.53-2. The nodalization shown in Figure 3.53-1 was maintained for all analyses performed using this model, with the dimensions adjusted to match specific embedments.

The overturning moment and vertical force induced by the LOCA were applied at the top of the skirt, which was constrained to remain plane and circular. The lateral force was similarly applied to the top of the skirt. The distribution of the applied lateral force was consistent with the shear distribution that would exist in a cylinder under flexure; it should be noted, however, that the analysis is insensitive to this assumed distribution. The prestress in the anchor bolts was modeled by including a preload in the appropriate truss elements. The reactions from the cavity wall were applied as concentrated nodal forces to the nodes corresponding to the edges of the wall.

Two sets of boundary conditions were applied. As noted, the top of the skirt was constrained to remain plane and circular in recognition of the stiffness of the reactor vessel to which it is fixed. At the basemat level, vertical springs were included to model the flexibility of the mat.

#### 2. Axisymmetric Embedment Model

The axisymmetric embedment model, complete with nodalization, is shown in Figure 3.53-3. For analysis of its response to an overturning moment a point force,  $V$ , was applied as shown. This force was prescribed to vary linearly with its normal distance to the reactor vessel centerline, using the harmonic loading option available in ANSYS. For analysis of its response to a lateral



force a horizontal force, H, was applied as shown. The distribution of this force about the embedment circumference was approximately that which would correspond to the shear distribution in a cylinder under flexure.

The boundary conditions imposed at the soffit of the basemat are described in full in section 2.2, Appendix K.<sup>2</sup> Boundary conditions appropriate to the axisymmetry of the model were applied to the basemat at the reactor vessel centerline. At the interface between the reactor vessel support skirt and the embedment (corresponding to the point of application of the vertical force in Figure 3.53-3), the rigidity of the skirt was included by prescribing that the interface remain plane. No restraint was imposed in the horizontal plane, however.

### 3. Embedment Substructure Model

The embedment substructure model is shown in Figures 3.53-4 and 3.53-5. This nodalization was maintained with the dimensions of the model varied to match specific embedments.

The vertical force induced by the overturning moment was applied to node 81 (see Figure 3.53-5). The beam representing the flange and sole plate distributed this load into the underlying concrete. Where shear anchors were present, the radial lateral force induced by the LOCA was included as point forces at the nodes corresponding to the shear anchor elevation. These point forces were in the direction of the applied load. Where shear anchors were not included, the lateral force was input in two parts: as distributed point forces on the concrete nodes beneath the sole plate to account for the friction, and as radial forces at node 125 to account for the load transferred to the vertical bearing plate via the sole plate. The cavity wall reactions were modeled as point forces at the three nodes on the top of the cavity wall stub.

The nodes on the underside of the pedestal were fixed.

Figure 3.53-1. Generic Embedment Model

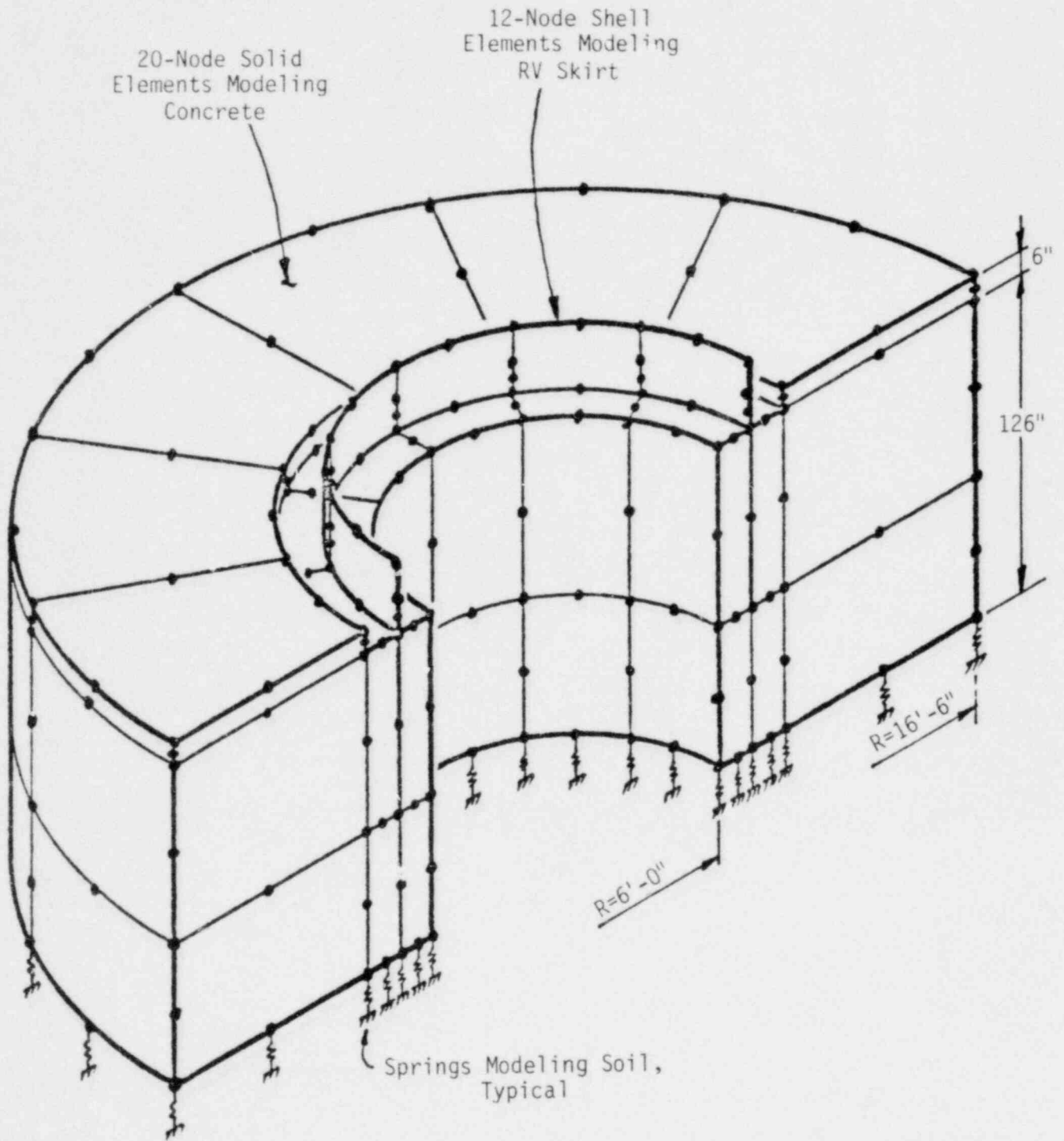


Figure 3.53-2. Generic Embedment Model Sections

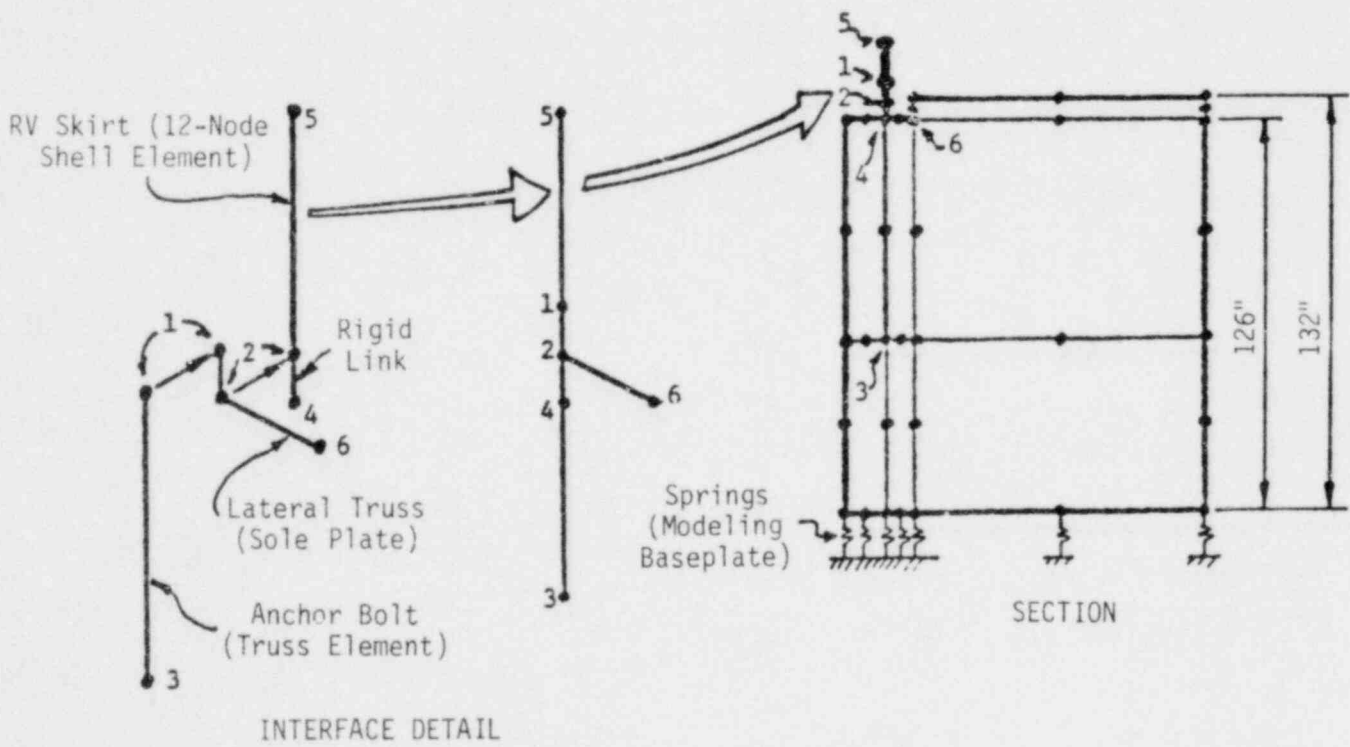
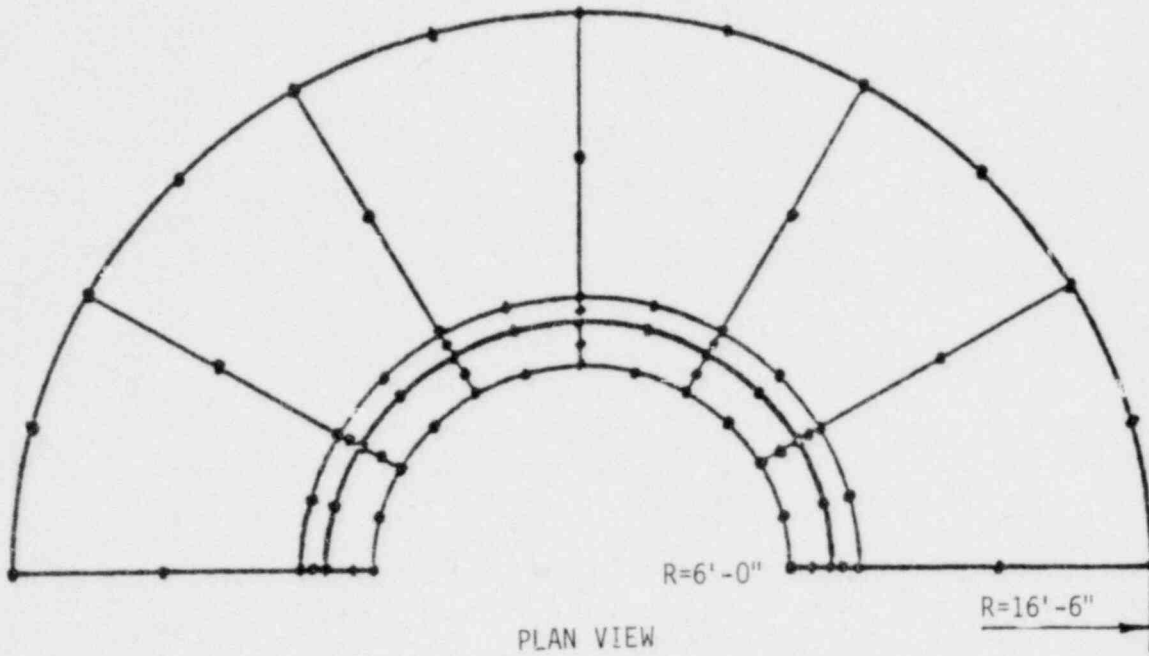


Figure 3.53-3. Axisymmetric Model of Embedment

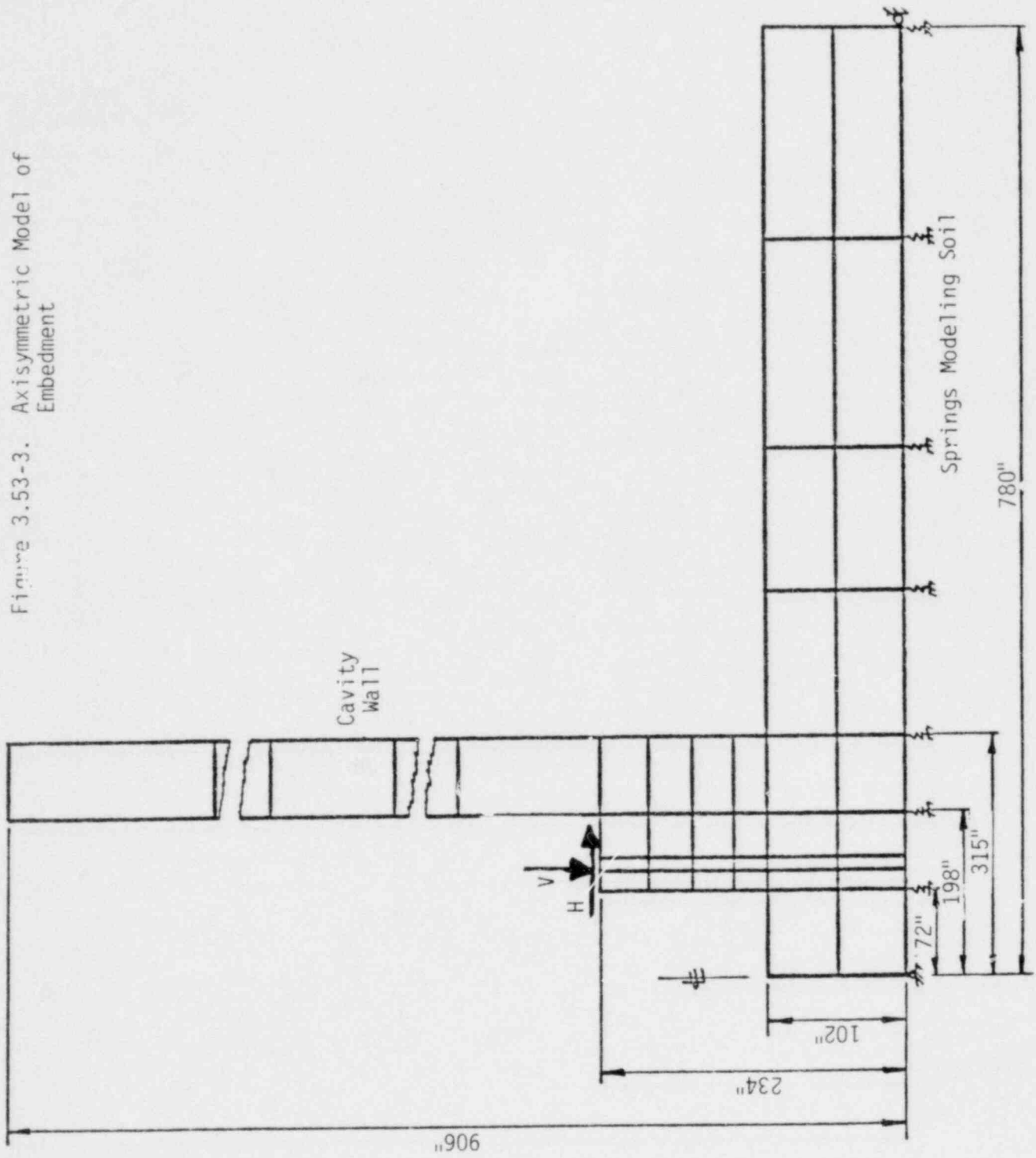
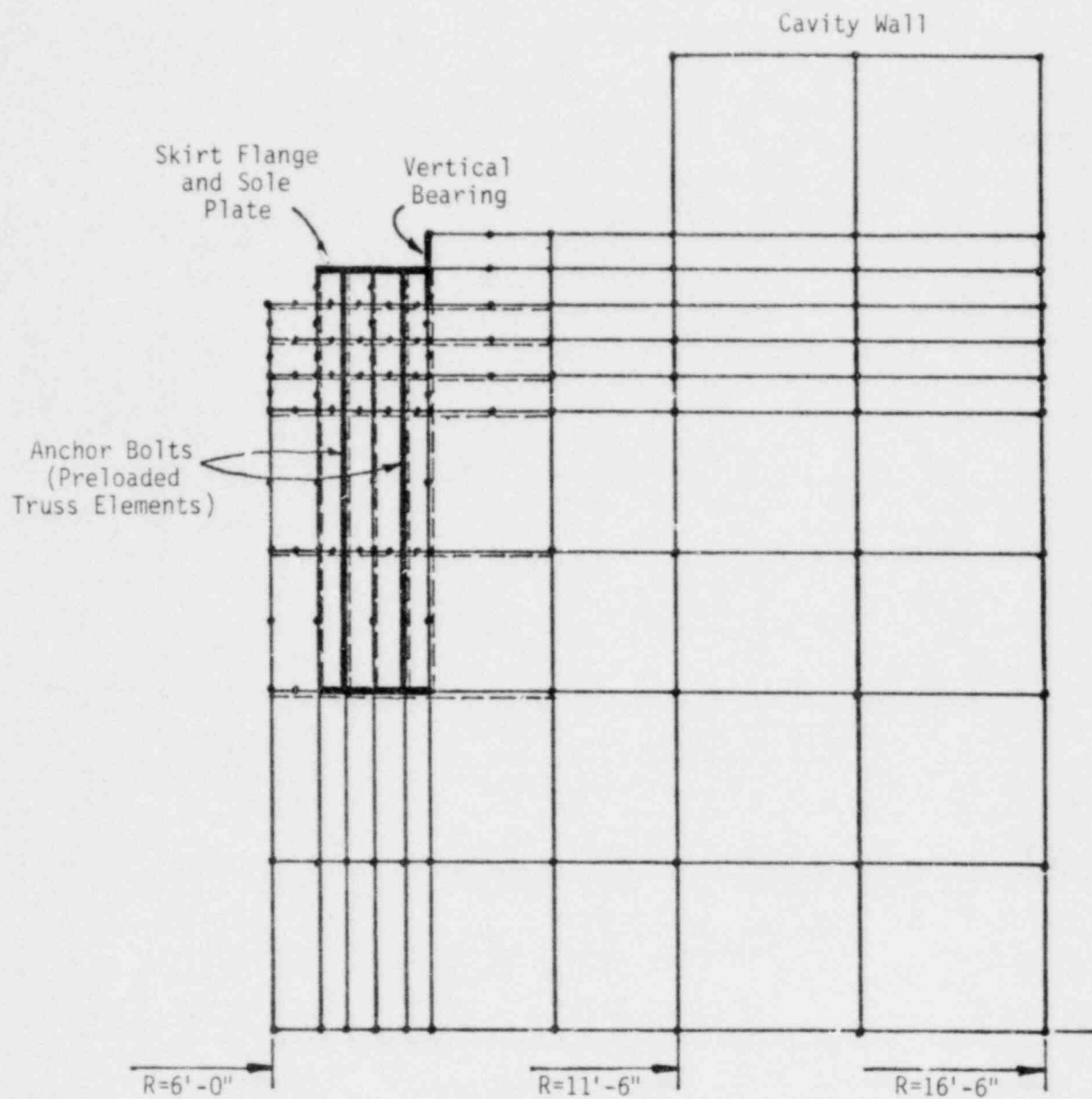
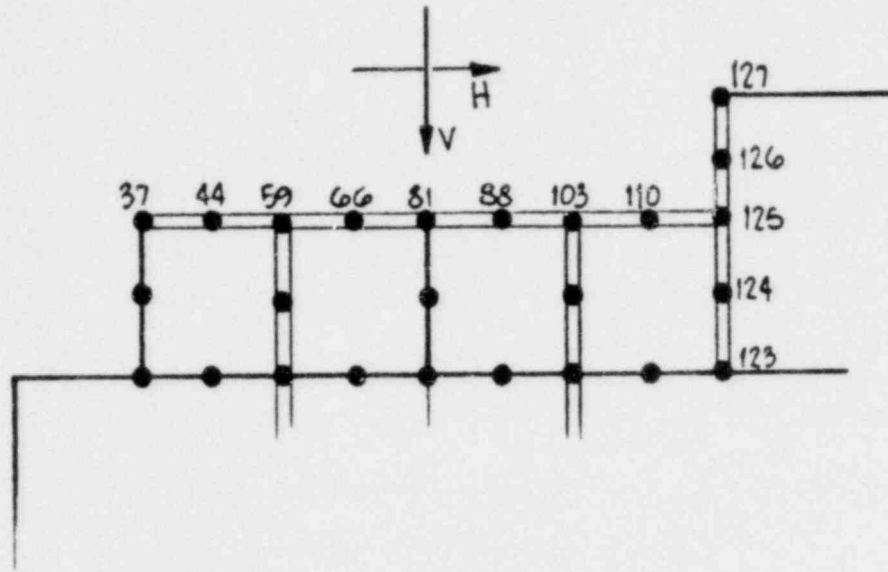


Figure 3.53-4. Embedment Substructure Model



Note: Dashed lines show reinforcing steel that was modeled.

Figure 3.53-5. Embedment Substructure Model Detail



3.54. Describe the differences between various plants. (Section 2.5, Appendix K)

### Response

Owing to the standard support skirt and reactor vessel configurations, there are no major differences between the reactor vessel support embedments for the skirt-supported plants. Minor variations in dimensions, details, and material properties are shown in Table 3.54-1. These are summarized below.

#### 1. Overall Dimensions

The depths of the reinforced concrete pedestals vary little, while their internal and external radii are constant. The depths of the basemats beneath the pedestals vary from 96 to 150 inches, but this will have no significant effect on the stiffness or strength of the pedestals.

#### 2. Connection of Skirt to Pedestal

The reactor vessel support skirt flange bears on a steel sole plate and against a circular vertical bearing plate, as shown in Figure K-2 of reference 2.

The dimensions of the skirt flanges do not change from plant to plant. The dimensions of the sole plates and the vertical bearing plates do change by relatively small amounts, and this was reflected in the plant-specific qualification. Details of the connections between the flange and the sole plate, the flange and the bearing plate, and the sole plate and the bearing plate also vary slightly, but this does not lead to significant variation among the responses of the various plants.

Three plants — Crystal River, Three Mile Island 1, and Three Mile Island 2 — have shear anchors beneath the sole plate. This was considered in the analysis of these plants were appropriate.

#### 3. Anchor Bolts

All anchor bolts are of 2-1/2-inch diameter with the exception of those for Oconee, which are of 2-inch diameter. The lengths of the anchor bolts vary from 65 to 109 inches. The prestress in the anchor bolts is (the least) 147 kips/bolt for Rancho Seco and (the most) 232 kips/bolt for the two Three Mile Island plants. The bolt strengths are relatively similar, with the exception of that for Rancho Seco, which is approximately 80% of the others. These variations in anchor bolt properties were considered where appropriate for both analysis and qualification.

#### 4. Pedestal Concrete Strengths

The minimum specified design strength of the pedestal concrete was 5500 psi for Arkansas and 5000 psi for the remaining plants. The 90-day strengths, which were used for qualification, varied from a low of 5500 psi for Arkansas to a high of 7455 psi for Three Mile Island 2. This variation in strengths had a direct impact on the qualification of the embedments.

#### 5. Openings in Pedestal

Variations between the plants in the orientation, size, and detailing of the incore instrumentation tunnels and access hatches are minor.

#### 6. Reinforcement

The pedestals of all the plants have different quantities and layouts of reinforcement. While all are well reinforced, those at Three Mile Island 1 and Crystal River are particularly strong. The Arkansas and Rancho Seco pedestals are the least reinforced, with the latter plant relying less on specific shear reinforcement.

In the Oconee and Arkansas plants, the reinforcement continuity is maintained across the liner plate beneath the pedestals. It is not on the remaining plants.

These variations were all taken into account for qualification. They have no effect on stiffness.



Table 3.54-1. Embedment Properties for Skirt-Supported Plants

	<u>ANO-1</u>	<u>Ocojee</u>	<u>CR-3</u>	<u>TMI-1</u>	<u>TMI-2</u>	<u>Rancho Seco</u>
<u>Overall Dimensions, in.</u>						
Pedestal depth	126	126	132	132	132	126
Basemat depth	108	102	150	108	138	96
<u>Skirt/Pedestal Connection</u>						
Sole plate width, in.	21	18	20	20	26	20
Sole plate thickness, in.	4	2.25	3.25	3.25	3.25	4
Vertical plate height, in.	11	12	12	12	12	7
Vertical plate thickness, in.	3	2	2.5	2.5	2	3
Shear anchors	No	No	Yes	Yes	Yes	No
<u>Anchor Bolts</u>						
Diameter, in.	2.5	2	2.5	2.5	2.5	2.5
Length, in.	65	70	92	92	92	109
Post-tension, kips/bolt	226	157	174	232	232	147
Yield strength, kips/bolt	130	130	140	130	130	105
<u>Concrete Strength, psi</u>						
Specified	5500	5000	5000	5000	5000	5000
90-day	5500	6355	7035	6650	7455	7275

3.55. Supply the gap dimensions. (Section 3.1.2, Appendix K)

Response

The gap dimensions for the Davis Besse 1 LOCA rings are given in Table 3.55-1. These gaps, between the pipe and the 12 shims which are located at regular angles on the inner face of each LOCA ring, apply for the hot condition. They are taken from the Bechtel Company's Davis Besse Nuclear Power Station Drawing No. C-196A, Revision 5.<sup>26</sup>

Table 3.55-1. Davis Besse LOCA Ring Gap Dimensions

Shim location <sup>a</sup>	Hot gap, in.	
	Hot leg	Cold leg
0°	0.125	0.125
30°	0.125	0.125
60°	0.125	0.125
90°	0.125	0.125
120°	0.166	0.166
150°	0.208	0.208
180°	0.250	0.250

<sup>a</sup>Shim location is the angle from top dead center of the pipe. Gaps are symmetric about the vertical centerline of the pipe.

3.56. Provide the basis for selecting stiffness values for the embedments from the parametric studies. (Section 4.1.1, Appendix K)

Response

The embedments of three plants were analyzed explicitly using three-dimensional finite element models. The three chosen were Rancho Seco, Oconee, and Crystal River 3. From these analyses, the parameters governing the stiffnesses of the embedments were identified. Using these parameters and the results for the Rancho Seco, Oconee, and Crystal River 3 embedments, stiffnesses (along with applicable elastic and ultimate limits) were calculated for the remaining plants.

For the rotational stiffnesses of the embedments prior to liftoff, the three plants were analyzed using the three-dimensional, generic embedment model. The pedestal depths and the anchor bolt length were identified as the governing parameters. For the rotational stiffnesses after liftoff, analyses were performed using the nonlinear embedment model. Linear response of the concrete was assumed, with the stiffnesses of the simplified concrete elements being chosen to reproduce the pre-liftoff stiffnesses determined earlier. The parameters governing the post-liftoff rotational stiffnesses were identified as the pedestal depth; the anchor bolt prestress, length, and diameter; and the pre-liftoff stiffness. Pre- and post-liftoff rotational stiffnesses for the remaining embedments were calculated accordingly.

The lateral stiffnesses of the three plants were determined from analyses using the generic embedment model. Based on the results of these analyses, pedestal depth was identified as the governing parameter, and stiffnesses for the remaining plants were determined accordingly.

The resultant variation between the rotational and lateral stiffnesses of the various plants was very small. The maximum difference was in the moments at which liftoff first occurred, and this was governed by the prestress in the anchor bolts.

Detailed descriptions of the generic and nonlinear embedment models are included in Appendix K of reference 2.

3.57. Seismic and LOCA evaluations do not exist for some of the components and piping analyzed. Provide these evaluations. (General)

Response

Seismic loadings were combined with LOCA for the evaluation of the reactor vessel supports and support embedments. Loadings and structural response data are found in sections 9 and 10 of reference 2.

As previously discussed with the staff during the Bethesda meeting on October 22, 1979, seismic loadings were not combined with LOCA in the evaluation of the remaining components, which included the fuel assemblies, control rod drives, core flood piping, primary piping, reactor internals, steam generator supports, and primary shield walls.

In combining seismic and LOCA for the vessel restraints and embedments, it was found that the influence of seismic loads on structural response was minimal when compared to LOCA. For this reason, it is judged that if seismic load were combined with LOCA for the remaining components, that no significant change in existing structural margin would result.

This is further substantiated in the application of the SRSS method for combining these loadings. When using this method, the seismic contribution to the total response has historically been very low (normally on the order of 1 to 2%). This further shows that the margins identified in section 10 of reference 2 for all components would not be significantly affected.

3.58. An evaluation of pump and steam generator supports is not provided for breaks occurring in the reactor cavity. Supply these evaluations.  
(General)

Response

Pump and steam generator supports were not evaluated for breaks within the primary reactor cavity.

For postulated breaks within the primary cavity, the jet impingement and cavity pressure effects on these supports would be essentially negligible because of the shielding effects of the primary cavity wall.

For this reason, postulated breaks external to the primary cavity were judged to be the design case for these supports, thereby eliminating the need to consider the postulated breaks within the primary cavity.

3.59. Some of the major system component evaluations were excluded from the evaluations for steam generator compartment pipe breaks. Provide these evaluations (include reactor internals). (General)

Response

The internals were evaluated only for the postulated breaks within the reactor subcompartment and for breaks at the first elbow outside the RV subcompartment. For other postulated breaks within the steam generator subcompartment, there would be no significant pressures within the reactor cavity, and no asymmetric loading component could be applied to the internals through their attachment to the reactor vessel.

The same is true for all components having a support function attached directly to the reactor vessel. These include not only the internals but the reactor vessel supports and embedments, fuel assemblies, control rod drives, core flood lines, and the service support structure.

Therefore, these components were excluded from the evaluation of breaks within the steam generator subcompartment since the postulated breaks within the reactor compartment were judged to be the design case.

ANO-1 - 1. Supply reactor vessel cavity wall drawings and include a detailed discussion of the critical loads, load paths, and functional integrity of the cavity wall. Particular attention should be focused on the location of the overstressed condition.

Response

The only area that is overstressed due to a 2.0A cold leg break is that immediately below the cold leg penetration. The model of the elements surrounding the cold leg penetration is illustrated in Figures ANO-1 - 1-1 and ANO-1 - 1.2.

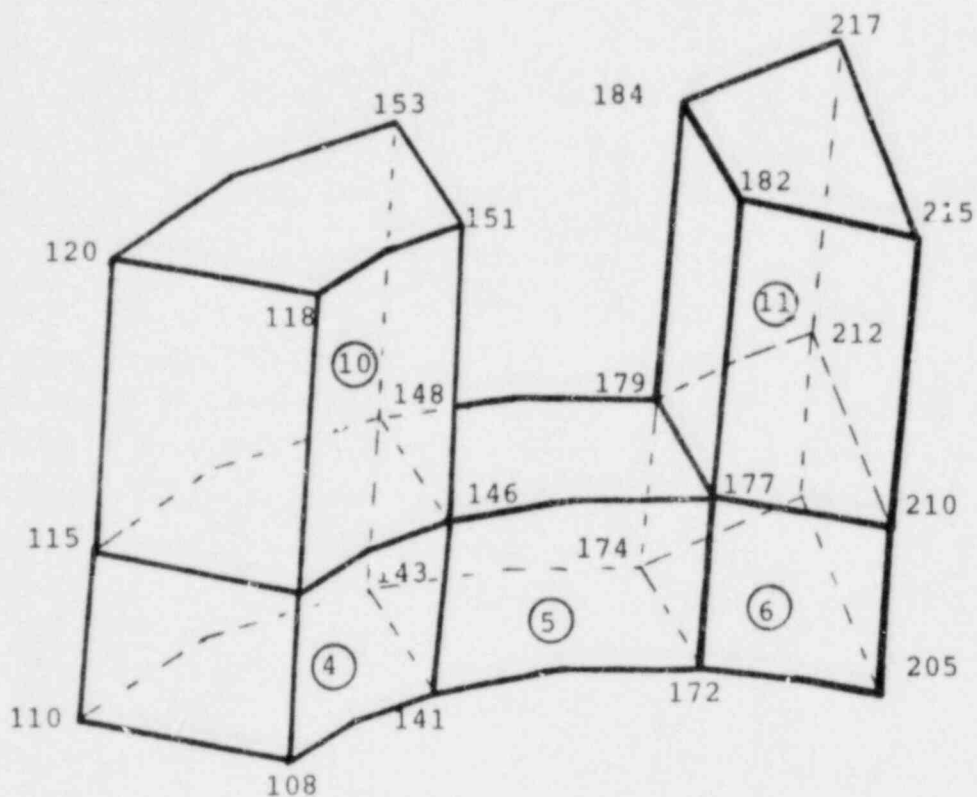
Figure ANO-1 - 1-2 shows the stress distribution on the elements surrounding the cold leg penetration. The overstress is seen to be very localized within the first element (element 5) directly below the cold leg. When the reinforcing steel yields, the load becomes distributed to the other hoop steel rebars surrounding the penetration. When the overstress in element 5 is distributed over the full height of the element, the average steel stress in that element is calculated to be 65.0 ksi. As shown in Figure ANO-1 - 1-2, the stresses on elements 4 and 6 are well below the allowable yield stress of 63 ksi. Thus, the overstress is localized to the area of element 5 and will not affect the functional integrity of the cavity wall.

Drawings of the ANO-1 cavity wall showing reinforcing steel are given as Figures ANO-1 - 1-3 and ANO-1 - 1.4 in Appendix A:

"Reactor Building Primary Shield Developed Elevation," Arkansas Nuclear One, Drawing C-169, Rev 3, Bechtel Corp.

"Reactor Building Primary Shield Reinforcing Plans," Arkansas Nuclear One, Drawing C-167, Rev 6, Bechtel Corp.

Figure ANO-1 - 1-1. Model of Elements Surrounding Cold Leg Penetration



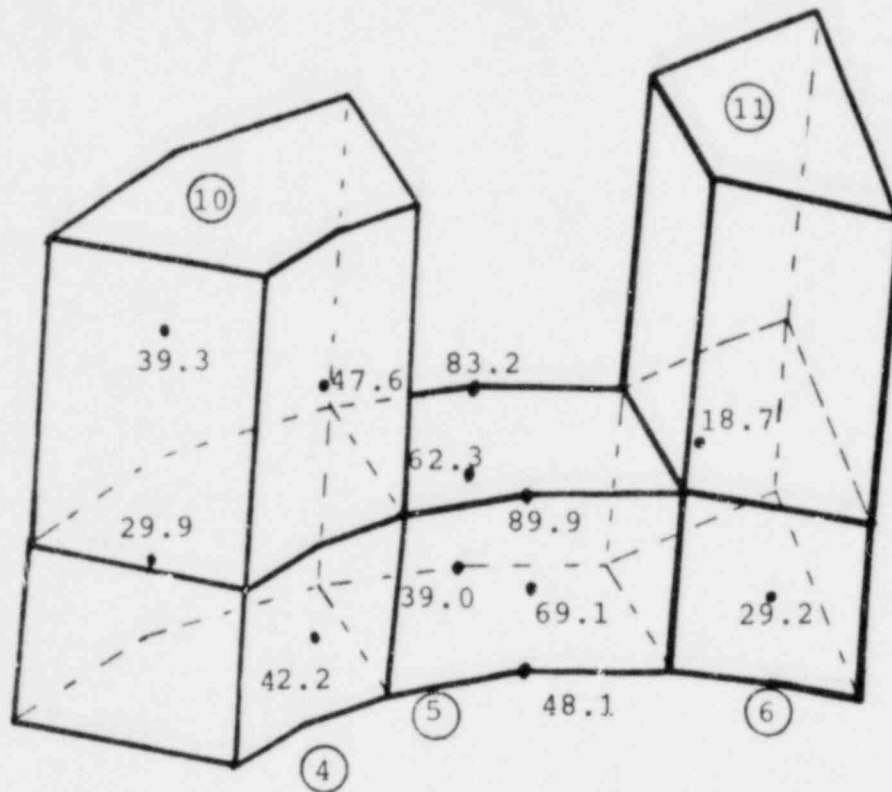
000 Corner node numbers

⓪ Three-dimensional element numbers

Note: Midside node numbers not shown.



Figure ANO-1 - 1-2. Stresses on Inner and Outer Element Faces Around Cold Leg Penetration (Combined Tension and Bending Stresses, ksi)



④ Element number

Note: Stresses on element 11 are lower than those on element 10.

Figures ANO-1 - 1-3 and ANO-1 - 1-4

(In Appendix A of this report)

ANO-1 - 2. Comment briefly on the response of restraints 2, 3, and 4 shown in Figure B-4. (Appendix B)

Response

For a hot leg guillotine break at the reactor vessel or at the first elbow, restraints 2, 3, and 4 shown in Figure B-4<sup>2</sup> are not impacted (also see Figure 4.1-1). That is, these breaks did not cause a hot leg displacement at restraints 2, 3, and 4 large enough to close the gap between the hot leg pipe and restraint. Table 9.8-5<sup>2</sup>, "Summary of Pipe Whip Restraint Loads for ANO-1," provides peak loads for restraint No. 1 and demonstrates that restraints 2, 3, and 4 are not impacted.

Figures ANO-1 - 2-1 and ANO-1 - 2-2 provide the location of hot leg restraints 1 through 4 (Mk-24, Mk-33, and Mk-34) and summarize the cold gap clearances.

Figure ANO-1 - 2-1. ANO-1 Hot Leg Restraints 1 and 2  
(From Bechtel Dwg. C-183)

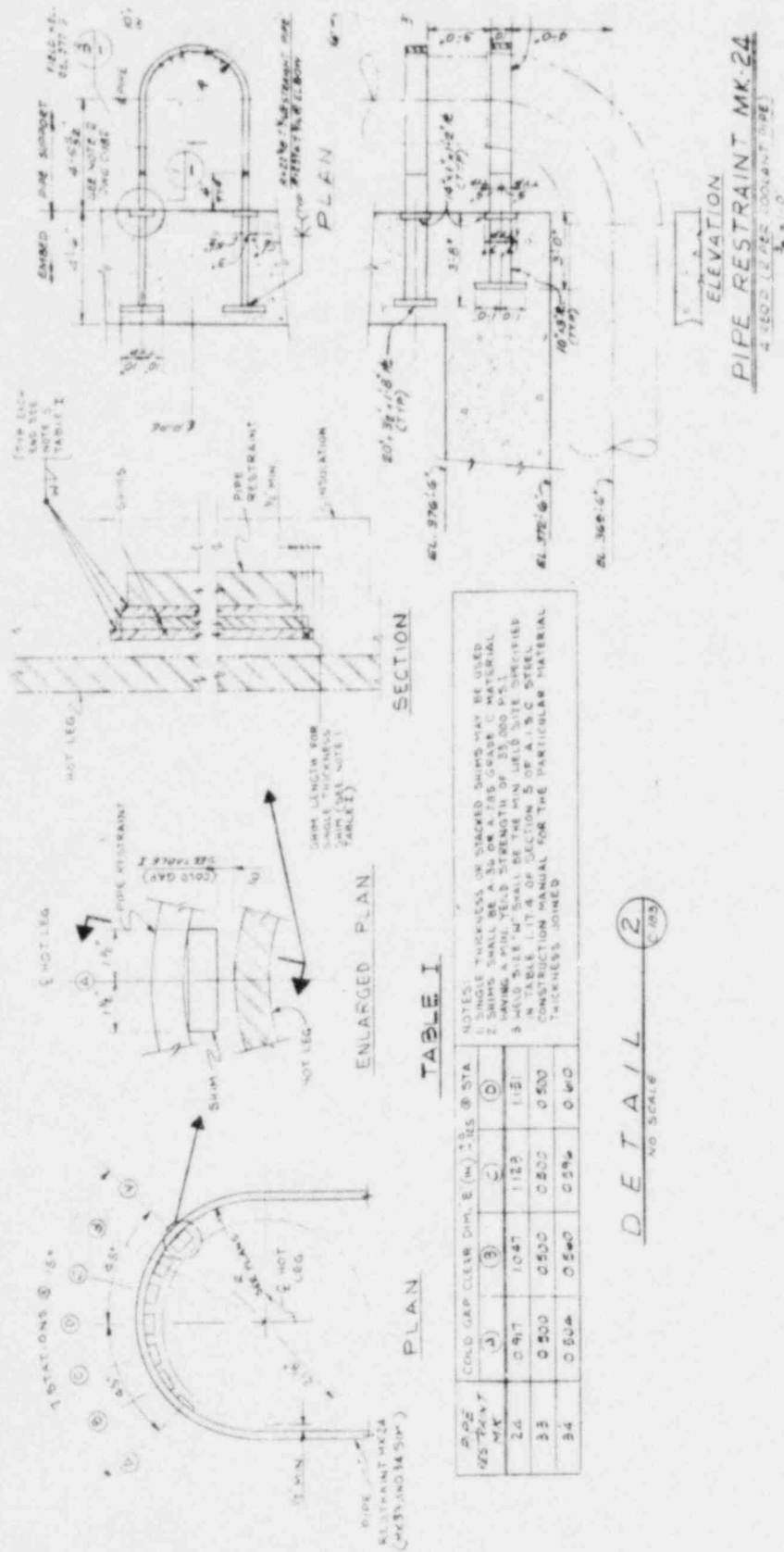
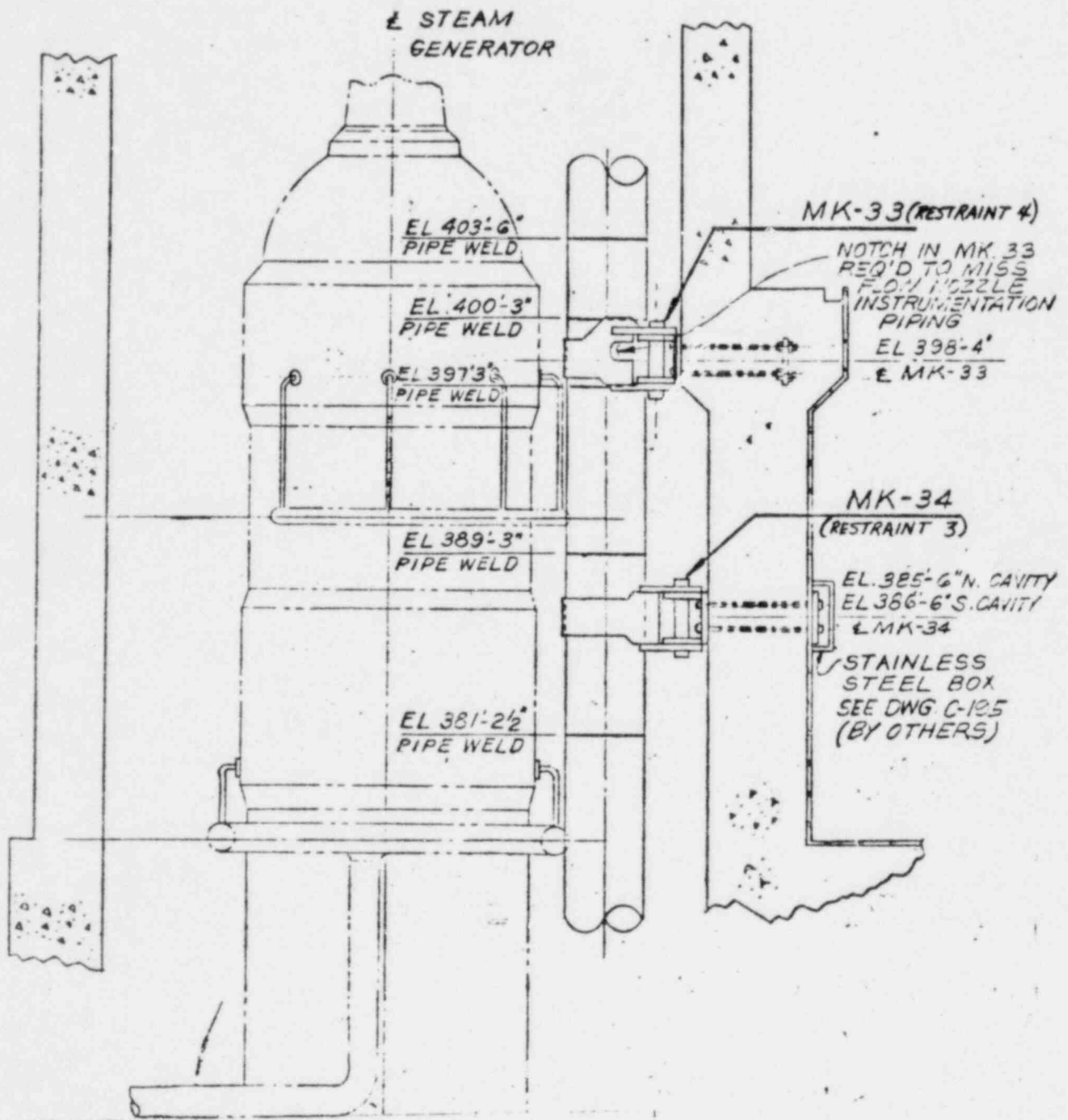


Figure ANO-1 - 2-2. ANO-1 Restraints 3 and 4



CR-3 - 1. Supply reactor vessel support and embedment drawings showing the general configuration. Provide a discussion of the loading and response of the embedment including the liner plate interface. Demonstrate that stability will be maintained and reactor shutdown and coolability will not be impaired.

#### Response

Only at the liner plate level were the stress limits defined by the acceptance criteria exceeded for the Crystal River embedment. The exceedance was only 4%, with a margin of at least 6% shown against the calculated ultimate strength.

In calculating the overturning load at the liner plate, a conservative interpretation of the load paths that would be induced by an asymmetric LOCA was adopted. The asymmetric LOCA produces moments of opposite direction on the reactor vessel support embedment and the cavity wall. These two structures are both linked to the basemat - the cavity wall through both the pedestal and the extremely strong secondary shield wall. This is most clearly shown in Figure I-1 of reference 2, which is reproduced here as Figure CR-3 - 1-1. If separation began to occur at the liner plate beneath the pedestal, displacement would be induced at the pedestal/cavity wall interface (shown as item 9 in the figure). This displacement would be resisted by the cavity wall, the reaction being transferred back to the basemat through the secondary shield wall. The resistance provided by the cavity wall was not considered in the Crystal River embedment analysis; all the overturning moment was assumed to pass into the basemat.

Thus, given the conservative manner in which the allowable stress at the liner plate was calculated, the small degree of overstress, and the additional resistance of the cavity wall, it is not considered that reactor vessel stability or coolability would be impaired by liner plate separation.

The following reactor vessel support and embedment drawings are provided in Appendix A.

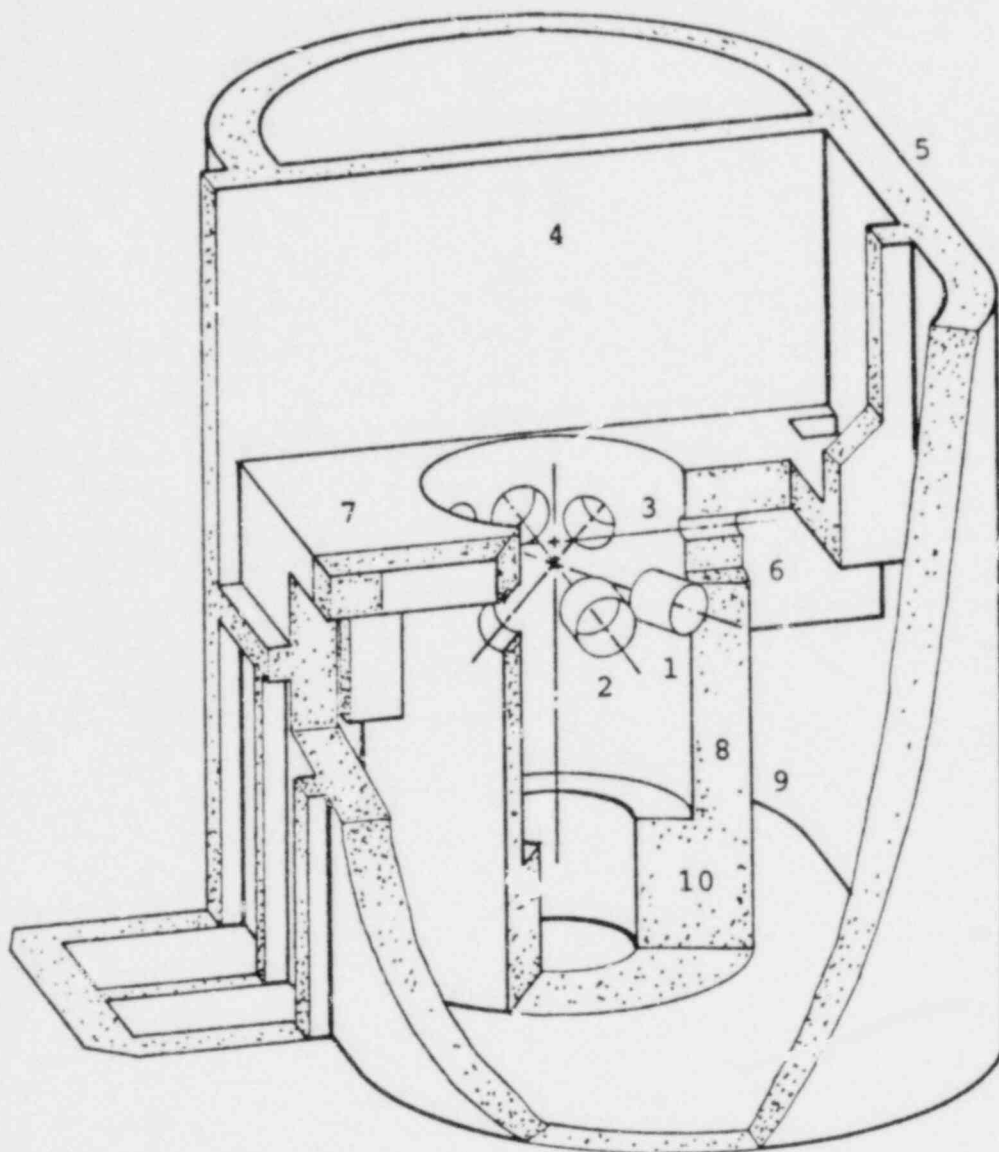
"Reactor Building - Steel, Reactor Vessel Support Anchor," Crystal River Plant, FPC Drawing S-521-022, Rev 6, Gilbert Associates, Inc.

"Reactor Building - Concrete Outline, Basement Floor, El. 95'," Crystal River Plant, FPC Drawing SC-421-008, Rev 5, Gilbert Associates, Inc.

"Reactor Building - Dowel and Anchor Bolt Plan, Basement Floor El. 95'," Crystal River Plant, FPC Drawing SC-421-009, Rev 1, Gilbert Associates, Inc.

"Reactor Building - Reactor Vessel Foundation, Plan and Sections," Crystal River Plant, FPC Drawing SC-421-015, Rev 4, Gilbert Associates, Inc.

Figure CR-3 - 1-1. Generic Cavity Wall



- 1 - Cold Leg Penetration
- 2 - Hot Leg Penetration
- 3 - Core Flood Line Penetration
- 4 - Fuel Canal Wall
- 5 - Secondary Shield Walls
- 6 - Core Flood Line Shield Wall
- 7 - Fuel Canal Floor
- 8 - Cavity Wall
- 9 - Pedestal - Cavity Wall Interface
- 10 - Pedestal



CR-3 - 2. Discuss in detail the modifications proposed for limiting the hot leg break area and explain what changes would be anticipated in component response. Supply drawings showing locations of the proposed modifications.

Response

Modification to limit the hot leg break area are not required. Section 9.8 of reference 2, which states that the CR-3 restraints were overstressed, is in error. Table 2-1<sup>2</sup> shows that the hot leg restraints are within acceptance limits; however, section 9.8 has not yet been revised to include these acceptable results obtained in subsequent analyses.

In addition, section 12.4<sup>2</sup> provides information that demonstrates that the reactor vessel support stability is maintained, thereby eliminating the need to modify the hot leg restraints to reduce the break area.

CR-3 - 3. Supply bracket plate and hot leg restraint drawings with a discussion of the applied loads responses and consequences of failure. Discuss in detail the calculation of the allowable loads.

Response

No response needed; see CR-3 - 2.

CR-3 - 4. Comment briefly on restraints 2 and 3 in Figure B-5. (Appendix B)

Response

For a hot leg guillotine break at the reactor vessel or at the first elbow, restraints 2 and 3 in Figure B-5 are not impacted (also see Figure 4.1-1).<sup>2</sup> That is, these breaks do not cause a hot leg displacement at restraints 2 and 3 large enough to close the gap between hot leg pipe and restraint. Table 9.8-4, "Summary of Pipe Whip Loads for Crystal River," provides peak loads for restraints 1, 2, and 3 shown in Figure B-5.

CR-3 - 5. Provide a detailed discussion of the load redistribution predicted to take place in the cavity wall. Supply drawings of the affected area and discuss the overall functional integrity of the wall.

Response

In the elastic analysis, moments at the cavity wall base exceed the allowable. This overstress is very localized around the cavity wall/pedestal interface. The values from the elastic analysis are only approximate, as the actual structure will behave in a nonlinear fashion. A plastic hinge forms around the base where overstress is predicted. The moment at these locations will not exceed the plastic moment for the cross section.

The excess moment is redistributed to the lower third of the cavity wall and causes increased hoop tension. Figures CR-3 - 5-1 and CR-3 - 5-2 show the distribution of hoop stress in the lower part of the cavity wall. Final stresses after the distribution of the excess moment are still below the allowable yield stress of 42 ksi for hoop stress.

The increase in hoop stress reported in Figure CR-3 - 5-2 was calculated for the maximum excess moment on the cavity wall. Actual moment distribution at various azimuths around the base of the wall is lower. The values reported are for the 1.092A hot leg break, which is the critical load case. Results for the 2.0A cold leg break are lower for all stress values.

The cavity wall retains functional integrity for all loading conditions.

Figure CR-3 - 5.3 is a drawing of the Crystal River 3 cavity wall with reinforcing steel. (Figure CR-3 - 5.3 provided in Appendix A)

"Reactor Building - Reactor Vessel Primary Shield Wall Plans and Sections," Crystal River Plant, FPC Drawing SC-421-012, Gilbert Associates, Inc.

Figure CR-3 - 5-1. Hoop Steel Stresses Before Redistribution of Excess Moment -  
 1.092A Hot Leg Break (0°), Steady-State Condition

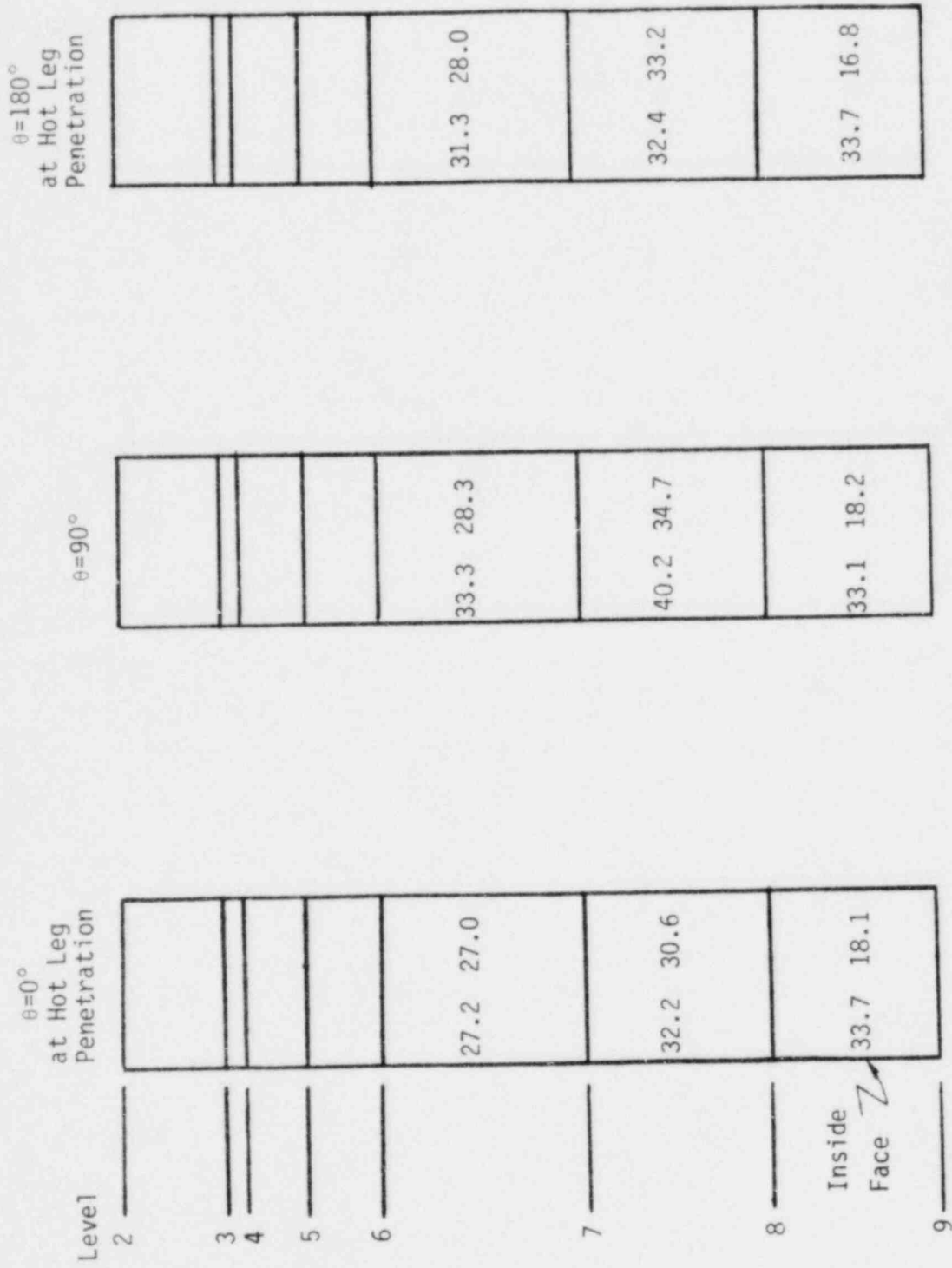
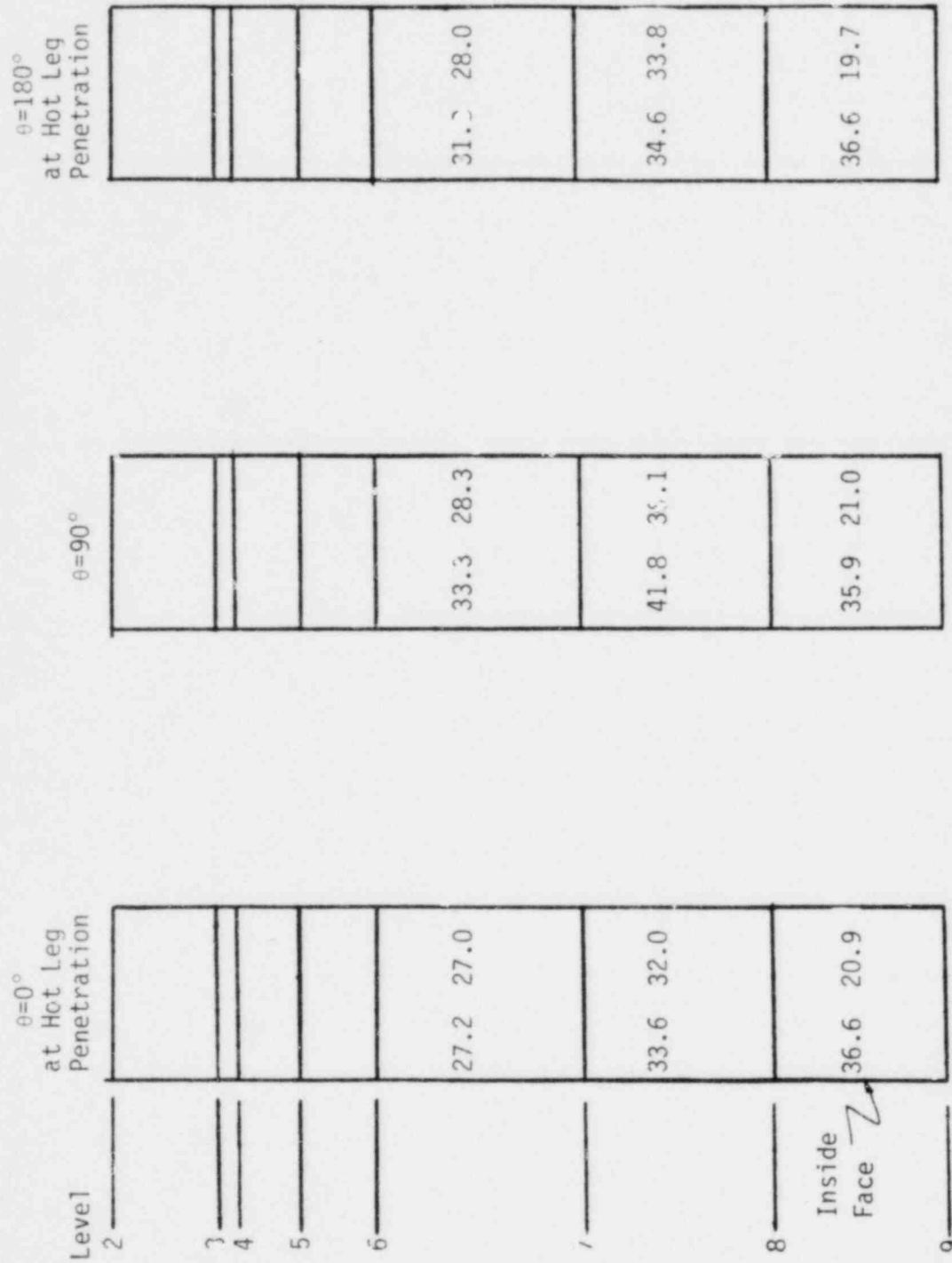


Figure CR-3 — 5-2. Hoop Steel Stresses After Redistribution of Excess Moment —  
 1.092A Hot Leg Break (0°), Steady-State Condition



DB-1 - 1. Detail the changes made to the skirt-supported models to facilitate the Davis Besse 1 analysis. Provide a detailed discussion of the response loads and the mathematical models, especially in the reactor vessel support regions.

#### Response

Davis Besse 1 is a nozzle-supported plant. During normal operating conditions, the reactor vessel (RV) is supported by four pairs of support beams, one pair at each cold leg. Each pair of support beams is connected by a support pad upon which the RV nozzle rests. There is also a LOCA ring at each hot and cold leg penetration. Figure K-3 of reference 2 shows a section at a cold leg penetration with the locations of a support beam and a typical LOCA ring. There are finite gaps between the LOCA rings and the pipes they encircle. Thus, the rings are not loaded under normal conditions.

During the LOCA event, the support beams take vertical loads (due to vertical and overturning loads on the RV) and lateral load. The lateral loads the support beams take are limited by the shear strength of the support pad/support beam bolts. After the support pad bolts fail in shear, the lateral load is taken by the LOCA rings. Thus, the LOCA rings at the unbroken hot and cold legs will be subject to horizontal, vertical, and out-of-plane (due to friction) loads.

A support beam model (Figures K-12 and K-13<sup>2</sup>) was developed for the analysis of the support beam and the concrete immediately surrounding it. This model was discussed in detail in section 3.2 of Appendix K.<sup>2</sup> The loadings applied to this model were the maximum vertical load due to RV deadweight plus LOCA and a lateral load, the magnitude of which was equal to the shear strength of the support pad/support beam bolts.

Two- and three-dimensional models of the LOCA rings were developed. These are fully described in Appendix K.<sup>2</sup> These models accounted for the interaction between the pipe and the restraint, the yielding of the LOCA ring, and the stiffness of the concrete in which it is embedded. The loading applied to these models was based on the assumption that all the lateral load was carried by the LOCA rings. An out-of-plane, frictional load equal to 42% of the resulting normal load on the LOCA rings was applied simultaneously. Neither the yielded support pad bolts nor these frictional forces were assumed

to act to help resist the lateral force on the RV. Clearly, this is conservative.

Loadings from the boundaries of these support beam and LOCA ring models were included in the analysis and qualification of the Davis Besse 1 cavity wall.



DB-1 - 2. Supply drawings of the reactor pressure vessel support system and the general plant layout.

Response

At the April 2-3, 1981 meeting with NRC and EG&G, this question was revised to state that the main area of interest was the LOCA ring. Figure DB-1 - 2-1 provides details of the hot and cold leg LOCA restraints, and Figure DB-1 - 2-2 is a general layout of the reactor pressure vessel support system including LOCA rings.

Figures DB-1 - 2-1 and DB-1 - 2-2

(In Appendix A of this report)

Oconee 1. Supply reactor vessel support and embedment drawings showing the general configuration along with a detailed discussion of the loading and structural response with particular attention to stability and component functions.

Response

As discussed at the April 2-3 1981, meeting in Bethesda, the question of the Oconee embedment stability is addressed in section 12.4 of reference 2. The following reactor vessel support and embedment drawings are attached:

"Reactor Building - Reactor Foundation Concrete," Oconee Nuclear Station Units 1, 2, 3; Drawing O-68A, Rev 15, Bechtel Corporation.

"Reactor Building - Reactor Foundation Reinforcing," Oconee Nuclear Station Units 1, 2, and 3; Drawing O-68B Rev 4, Bechtel Corporation.

"Reactor Building - Primary and Secondary Shield walls, Plans at El. 777'-6" and 802'-0, Concrete," Oconee Units 1, 2, and 3; Drawing O-69A, Rev 12, Bechtel Corporation.

"Reactor Building - Primary and Secondary Shield Walls, Sections and Elevations, Concrete," Oconee Units 1, 2, and 3; Drawing O-69D, Rev 15, Bechtel Corporation.

Ocone 2. Supply drawings of the control rod drive mechanism and housing and discuss the applied loads, responses, and modified mechanisms.

Response

Drawings of the control rod drive mechanism and housing are found in Figures Ocone 2-1 through 2-3.

The applied loadings are found in section 9.4 of reference 2 and the structural response information is contained in section 10.4 of reference 2.

For a discussion of the modified mechanism please refer to BAW-10047, Rev 1.<sup>13</sup>

OCONEE 1, 2

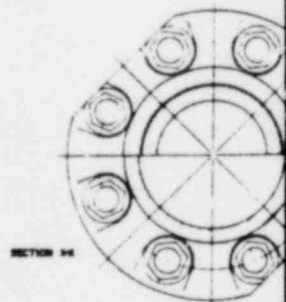
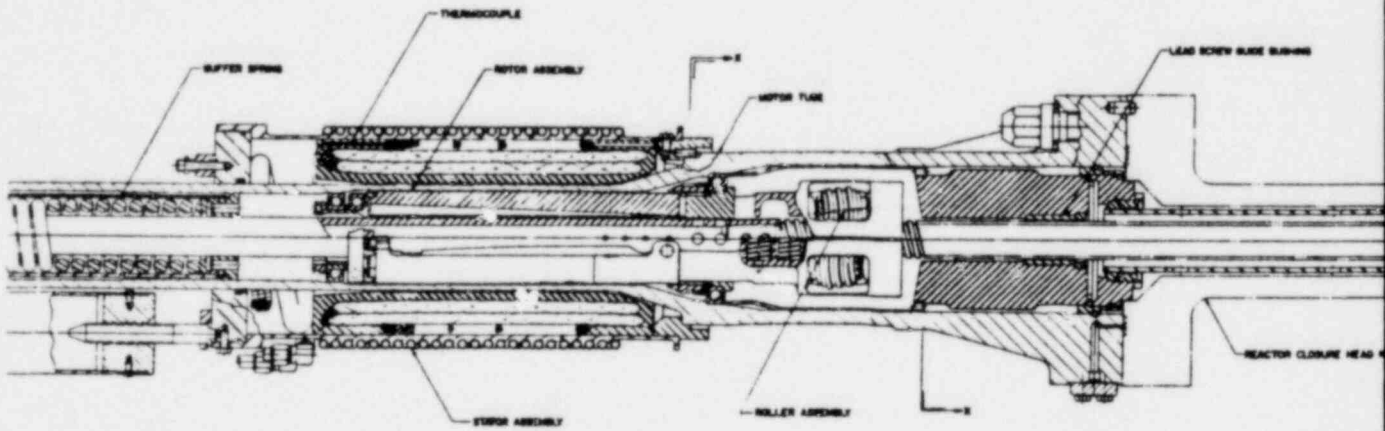
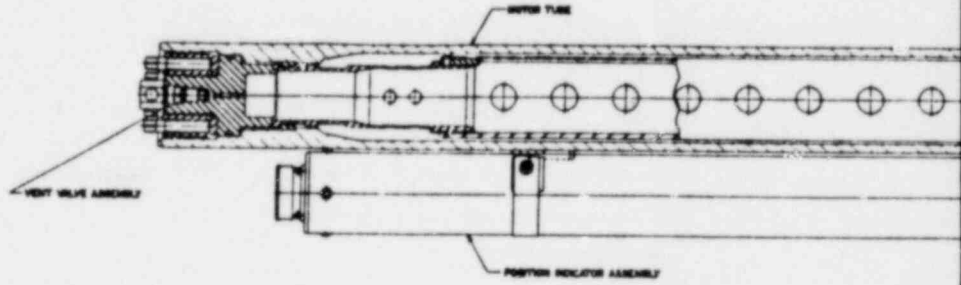
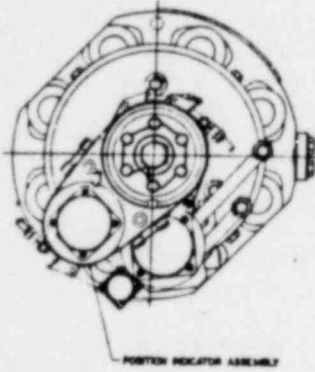
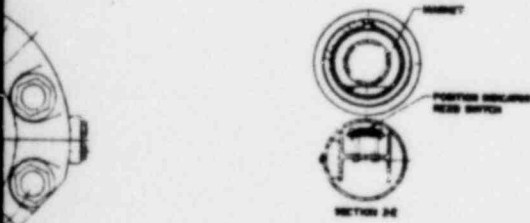
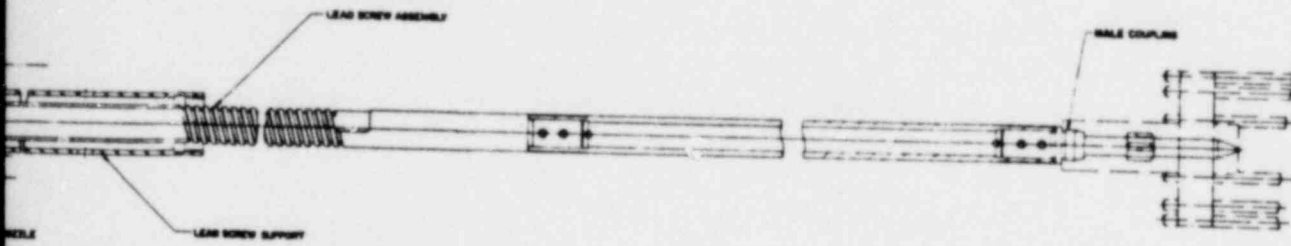
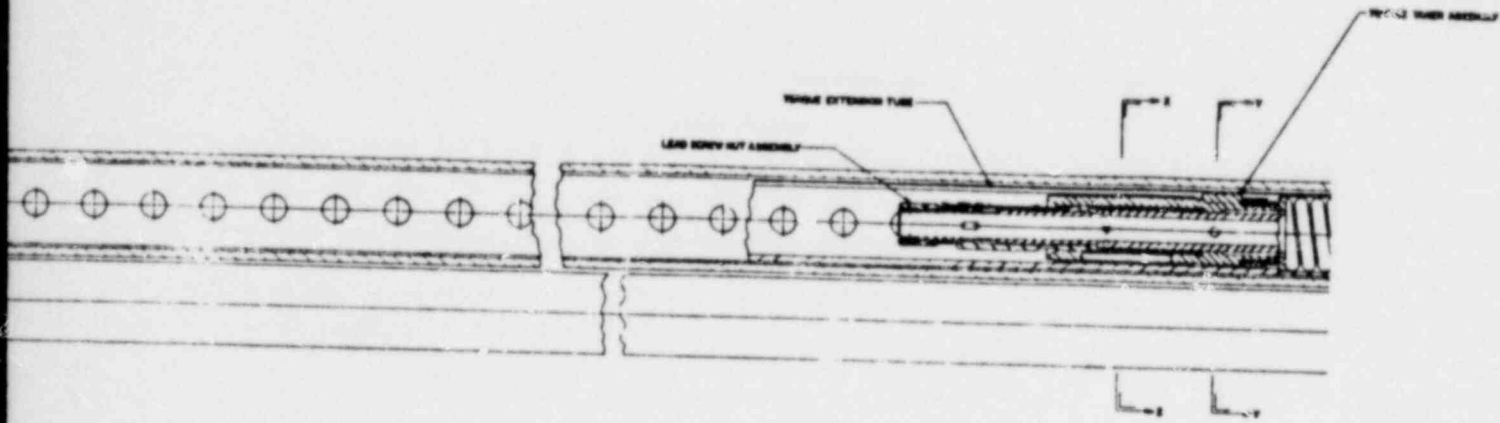
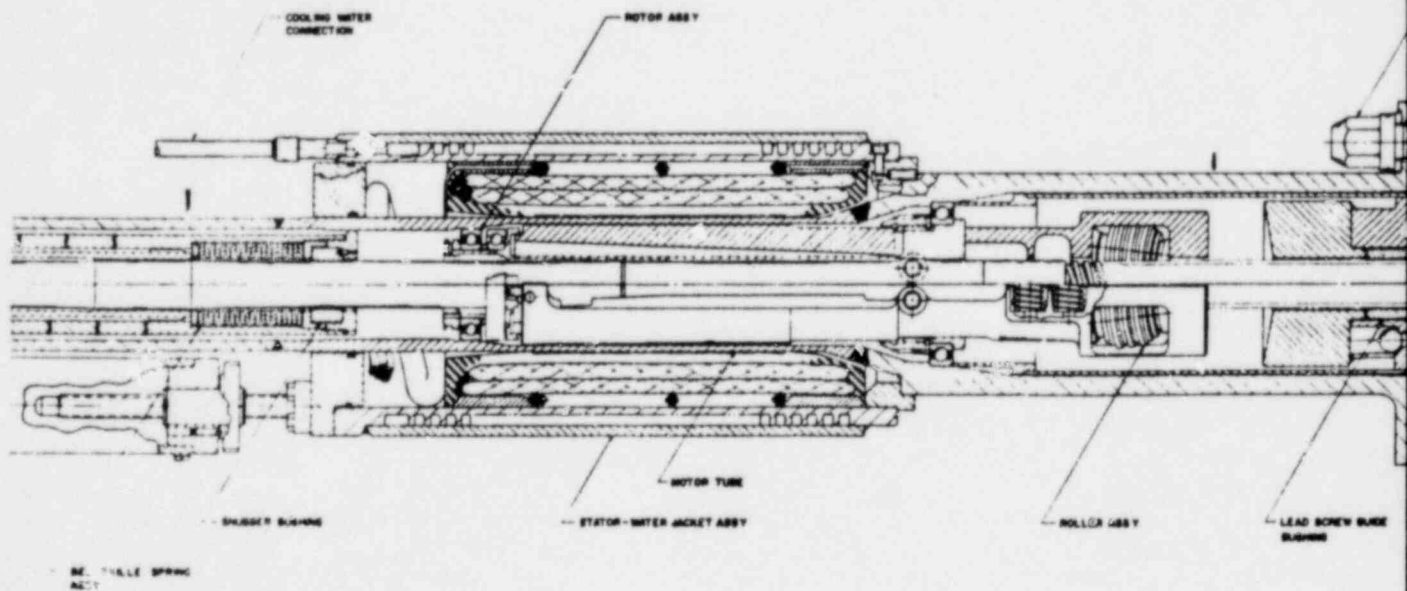
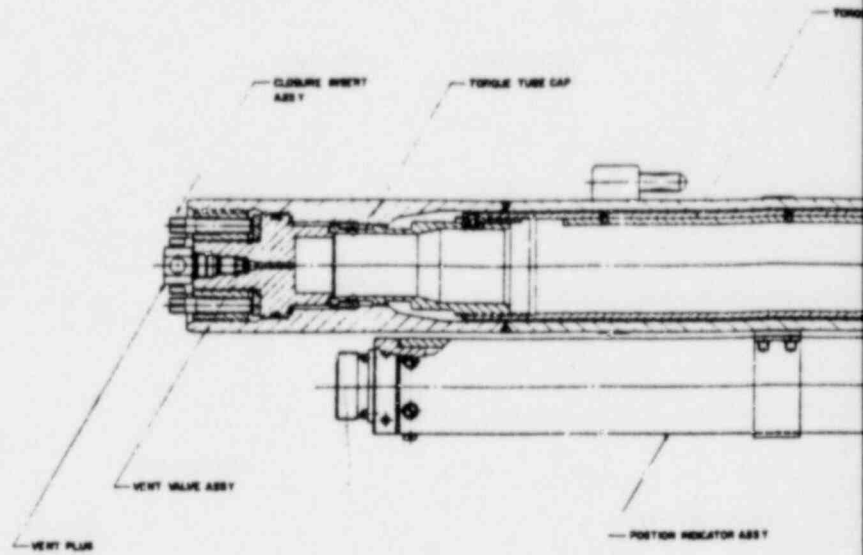
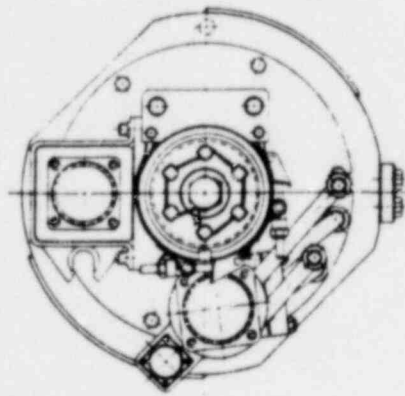


Figure Ocone 2-1. Control Rod Drive Mechanism





BE. VALVE SPRING ASST

Figure Ocone 2-2. Control Rod Drive-  
Vertical Section

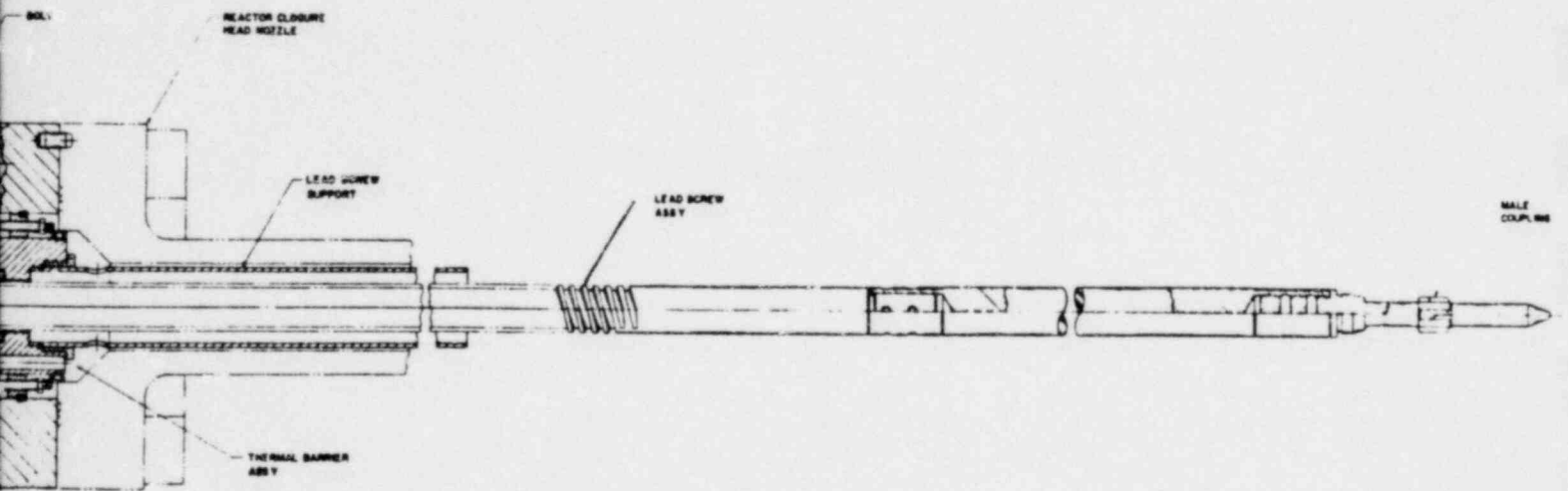
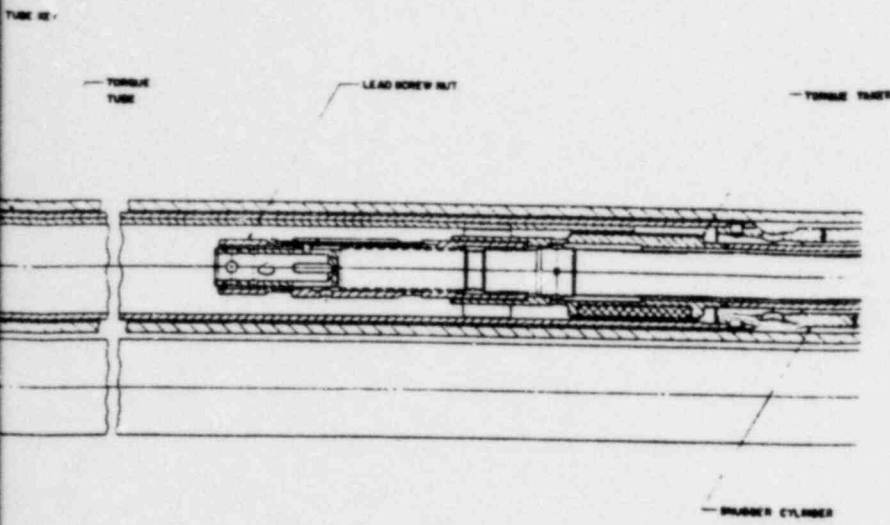
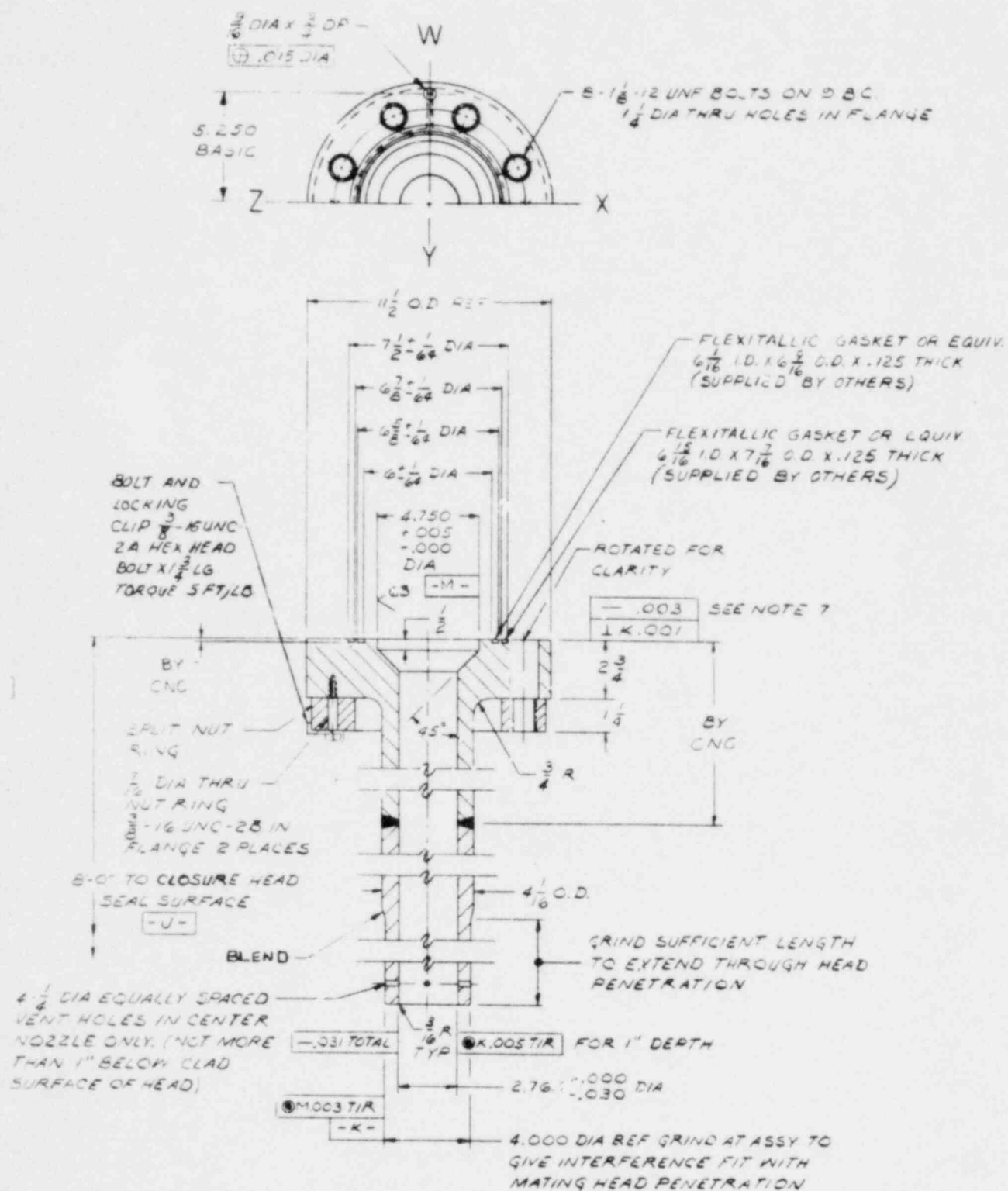




Figure Ocone 2-3. Control Rod Drive RV Head Nozzle



Rancho 1. Supply drawings of the reactor vessel embedments detailing the location where the acceptance criterion is exceeded. Discuss the various responses and the effect on overall stability.

Response

The peak asymmetric LOCA loading leads to exceedance of the acceptance criteria in three locations for the Rancho Seco reactor vessel support embedment.

In two of these locations, above the incore instrumentation hatch and around the vertical hatch, the allowable shear stress is exceeded locally by 8%.

The actual shear stress is still only 91% of the calculated ultimate. The allowable shear stress was calculated using the relevant code procedures.<sup>23</sup> Given the intensity of the reinforcement and the depth of the pedestal, this is considered to be a conservative methodology. Furthermore, the reinforcement will ensure that any failure will be relatively ductile, leading to a considerable redistribution of the shear stress to neighboring, less highly stressed locations. This effect was not taken into account.

In the third location, against the vertical bearing plate, the allowable compressive stress on the concrete was exceeded by 7%. The actual stress was, however, only 75% of the calculated ultimate. Again, conservative code procedures were used to calculate this allowable. Given the increases in allowables that were obtained for the concrete under the sole plate when explicit nonlinear analysis was performed, this 7% overstress is not considered significant. Furthermore, no account was taken for redistribution.

In all three locations, the overstress occurs locally at an opening in the pedestal. Only one of the six possible LOCAs would load the pedestal in this manner. Given the local nature of the calculated overstresses, the conservative manner in which the allowables were calculated and the results of analyses described in section 12.4<sup>2</sup>, it is concluded that neither reactor vessel stability nor component functionality will be adversely affected.

The following reactor vessel support and embedment drawings are attached as requested:

"Reactor Building - Area 1, Reactor Support," Rancho Seco Nuclear Station - Unit 1, Drawing C-368, Rev 7, Bechtel Corporation.

"Reactor Building - Area 1 Reinforced Concrete Primary Shield - Sheet 1," Rancho Seco Nuclear Station - Unit 1, Drawing C-309, Rev 2, Bechtel Corporation.

"Reactor Building - Area 1 Reinforced Concrete Primary Shield - Sheet 2,"  
Rancho Seco Nuclear Station - Unit 1, Drawing C-310, Rev 1, Bechtel Corpora-  
tion.

"Reactor Building - Area 1 Reinforced Concrete Primary Shield - Sheet 3,"  
Rancho Seco Nuclear Station - Unit 1, Drawing C-311, Rev 4, Bechtel Corpora-  
tion.

"Reactor Building - Area 1 Reinforced Concrete Primary Shield - Sheet 4,"  
Rancho Seco Nuclear Station - Unit 1, Drawing C-312, Rev 4, Bechtel Corpora-  
tion.

Rancho Seco 2. Comment briefly on restraints 1 and 4 in Figure B-6.

Response

For a hot leg guillotine break at the reactor vessel or at the first elbow, restraints 1 and 4 in Figure B-6<sup>2</sup> are not impacted (also see Figure 4.1-1<sup>2</sup>). That is, these breaks do not cause a hot leg displacement at either restraint 1 or 4 large enough to close the gap between hot leg pipe and restraint. Table 9.8.6<sup>2</sup>, "Summary of Pipe Whip Loads for Rancho Seco," provides peak loads for restraints 1, 2, 3, and 4 shown in Figure B-6<sup>2</sup>.

- TMI-1 - 1. Discuss in detail the differences between the TMI-1 and CR-3 power plants in the following areas:
- a. Reactor vessel supports.
  - b. Reactor vessel support embedments.
  - c. Control rod drive mechanism housings.
  - d. Proposed modifications for limiting the hot leg break area.

Response

For TMI-1 and CR-3 there are no significant differences in the reactor vessel supports or the control rod drive mechanism housings.

There are no major differences between the reactor vessel support embedments for the CR-3 and TMI-1 power plants. The density and layout of the reinforcements in the two embedments are relatively similar. Major dimensions, together with anchor bolt properties, are summarized in Table TMI-1 - 1. Differences between the plants are the following:

- The 90-day concrete strength was 6% higher for the CR-3 plant.
- A number of small differences in detail do exist, and these affect the relative strength ratios reported for the two embedments:
  - The CR-3 plant has shear anchors beneath the sole plate, whereas TMI-1 does not.
  - More horizontal load is thrust against the vertical bearing plate in the TMI-1 plant than at Crystal River.

A nonlinear vessel stability evaluation was performed for CR-3 (section 12.4, reference 2) which demonstrated that the vessel remained stable for the postulated hot leg LOCA. For this reason CR-3 does not plan to install an additional hot leg restraint in the penetration.

As noted in reference 14, TMI-1 plans to initiate a program to evaluate the probability of leak before break and/or conservatisms in the Phase II analysis to determine whether an additional hot leg restraint is required. Pending the outcome of the program, they will, by the first refueling outage, either install a restraint or provide justification for the adequacy of the existing plant design.

Table TMI-1 - 1-1. TMI-1 and CR-3 Embedment Properties

<u>Property/parameter</u>	<u>CR-3</u>	<u>TMI-1</u>
Pedestal depth, in.	132	132
Basemat depth, in.	150	108
Anchor bolt length, in.	92	92
Anchor bolt diameter, in.	2.5	1.5
Effective post tension, kips/bolt	174	232
Bolt yield stress, psi	140	130
90-day concrete compressive strength, psi	7035	6650

TMI-1 - 2. Supply hot-leg pipe whip restraint drawings and discuss analytical results with respect to these drawings. Detail the consequences of failure of this support.

Response

For Crystal River, the hot leg restraint did not exceed allowable stresses for the postulated hot leg LOCA, and vessel stability was demonstrated. (See response to question CR-3 - 2.)

As noted in reference 14, TMI-1 will evaluate the probability of leak before break and/or analytical conservatisms and will either modify and/or install restraints by the first refueling outage or will provide justification for the adequacy of the existing plant design. (See response to question TMI-1 - 1.)

TMI-1 - 3. Comment briefly on restraint 2 in Figure B-2.

Response

For a hot leg guillotine break at the reactor vessel or at the first elbow, restraint 2 shown in Figure B-2<sup>2</sup> is not impacted (also see Figure 4.1-1<sup>2</sup>). That is, these breaks do not cause a hot leg displacement at restraint 2 large enough to close the gap between hot leg pipe and the restraint. The "Summary of Pipe Whip Loads for TMI-1," Table 9.8-2<sup>2</sup>, provides peak loads for restraints 1 and 2 shown in Figure B-2.



TMI-1 - 4. The following information is requested on the reactor vessel cavity wall:

1. Drawings describing the wall and the surrounding structures.
2. Allowable values based on the wall primary functions.
3. A discussion of the load distribution, load path, and magnitude along with a comparison of the results to a functional allowable.
4. A definitive margin of safety for the wall.
5. Definitive ratio of ductility as used in this context.

Response

1. The requested drawings are included as Figures TMI-1 - 4-1, TMI-1 - 4-2, and TMI-1 - 4-3 in Appendix A.

"Sections A-A and B-B, Reactor Building," Three Mile Island Nuclear Station, Unit 1 Drawing E-002-004, Rev 33, Gilbert Associates, Inc.

"Reactor Building - Concrete Reactor Vessel Primary Wall, Elev. 289'-7" to Elev. 317'-0"." Three Mile Island Nuclear Station, Unit 1, Drawing E-421-016, Rev 6, Gilbert Associates, Inc.

"Reactor Building - Concrete Plan Floor Elev. 346'-0"," Three Mile Island Nuclear Station, Unit 1, Drawing E-421-009, Rev 6, Gilbert Associates, Inc.

2. The functional integrity of the wall is maintained for a given load if the maximum tensile strain in the reinforcement is less than 3.75%. This corresponds to an average maximum pressure of 280 psi over the cavity wall and a maximum radial displacement of 4.6 inches.
3. The analysis was performed for a 1.295A hot leg break. At maximum load, the concrete has cracked over much of the cavity wall and pedestal. At this load, the total load applied to the cavity wall is resisted as follows:
  - a. Hoop reinforcement, 57%
  - b. Pedestal, 25%
  - c. Fuel canal slab and wall system, 18%

Yielding of the hoop reinforcement extends over the full length of the cavity wall except the hoop steel in vicinity of the fuel canal slab and pedestal. Maximum strain of the hoop reinforcement is 1.27%.

Yielding of the vertical reinforcement primarily extends over three regions:

- a. At the outside face of the cavity wall, the reinforcement experiences tensile yielding for a distance of about 12 feet below the nozzle belt region.
- b. At the inside face of the cavity wall, at the pedestal/cavity wall interface, yielding extends from 1.5 feet above the pedestal to 5 feet below it.
- c. At the inside face of the cavity wall, at midheight, there is localized compressive yielding. The strain of the reinforcement in this area is less than 1.42%.

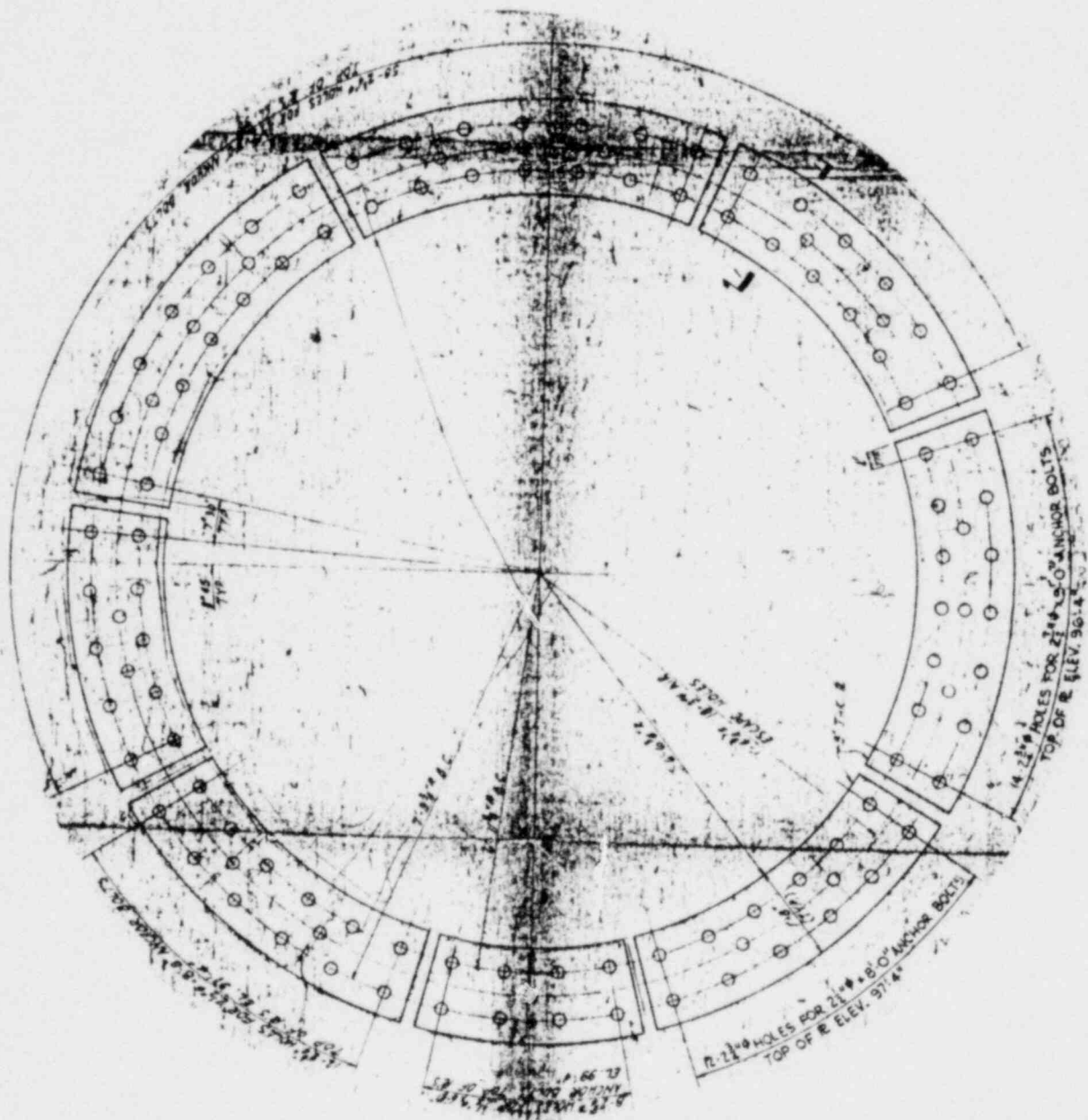
At no point is steel strain over 1.42%, or wall displacement over 2 inches. These values are below the functional allowables.

4. The average pressure that produces 3.75% strain is 280 psi. The maximum average pressure for a 1.295A hot leg break is 255 psi. Therefore, the margin of safety for the wall is 1.10.
5. The maximum radial displacement at the start of the yielding of the hoop reinforcement is 0.24 inch. The corresponding displacement at a pressure of 255 psi is 1.91 inches. Therefore, the ratio of ductility is 7.96. Results were extended from the original analysis for actual peak loads (255 psi) to ultimate loads (280 psi) as discussed in reference 27. From these data, the corresponding displacement for a pressure of 280 psi is 4.60 inches, which gives a ratio of ductility of 19.2. Therefore, the ratio of actual ductility to the function level ductility is 0.41.

APPENDIX A  
Drawings



- NOTES CONTINUED:
12. A CERTIFIED ENGINEER SHALL BE SUBMITTED
  13. TYPICAL BRACKETING
  14. 1/2" DIA. WALL THICKNESS
  15. 1/2" DIA. WALL THICKNESS
  16. 1/2" DIA. WALL THICKNESS
  17. 1/2" DIA. WALL THICKNESS
  18. 1/2" DIA. WALL THICKNESS
  19. 1/2" DIA. WALL THICKNESS
  20. 1/2" DIA. WALL THICKNESS

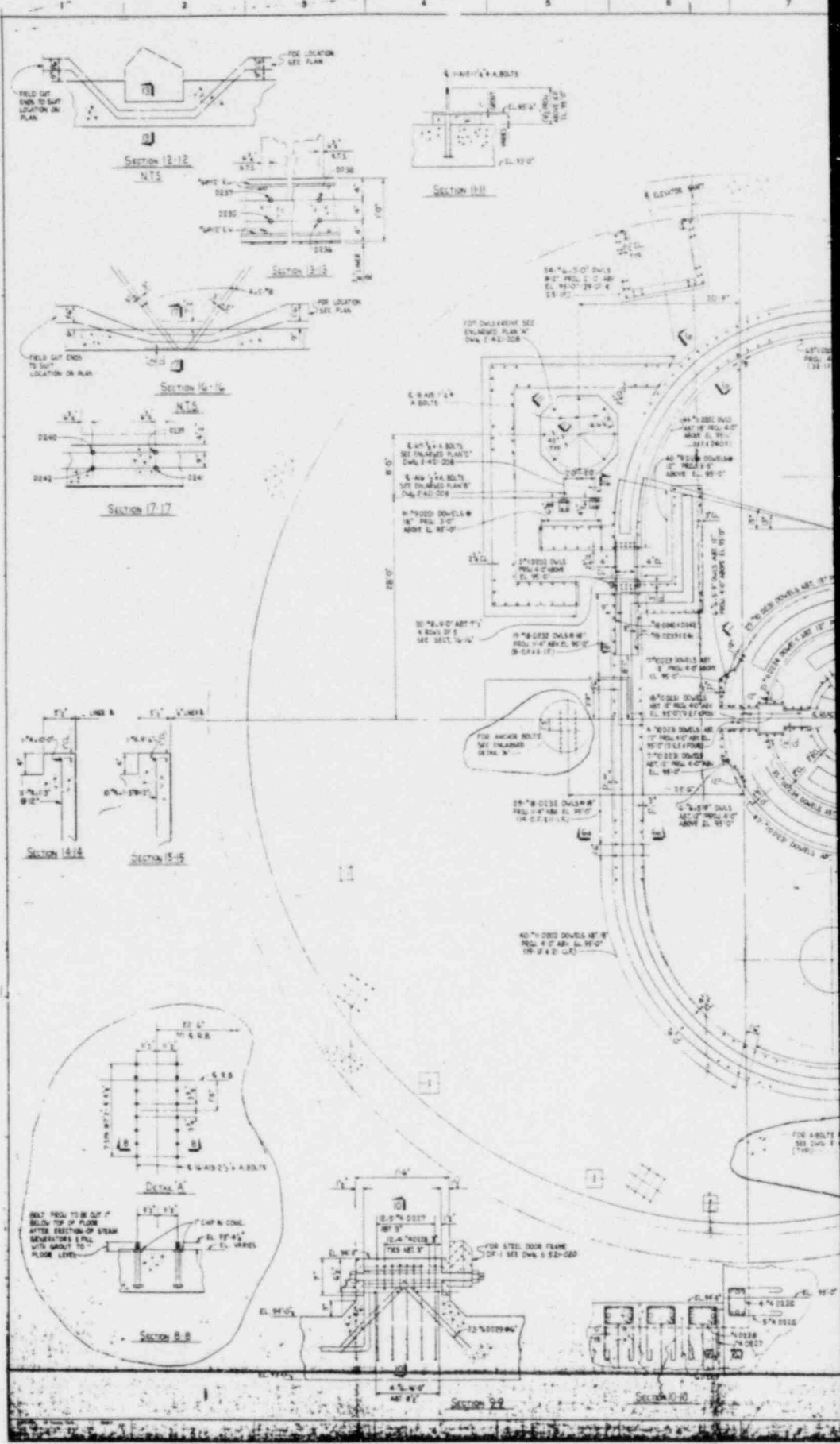


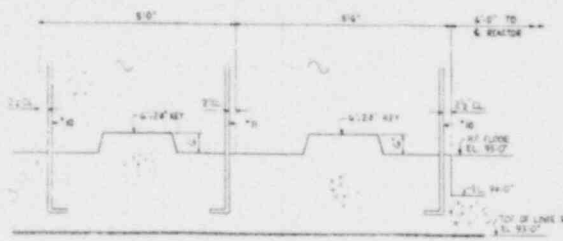
PLAN 304 - 2377  
EL. 104 - 07

PLAN BOTTOM ANCHOR BOLTS

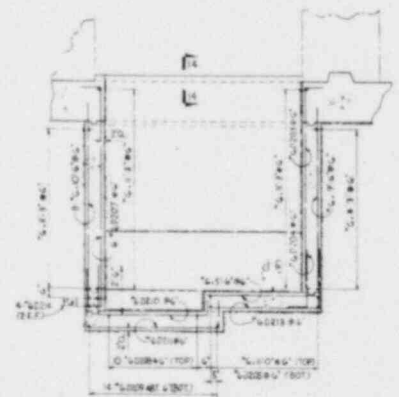
2 1/2" DIA. ANCHOR BOLTS (AREA IN FIG. 18)  
1/2" THICKNESS

<b>FLORIDA POWER CORPORATION</b>	
37 PETERSBURG, FLORIDA	
CRYSTAL RIVER PLANT	
UNIT NO. 3	855,000 KW
REACTOR BUILDINGS - STEEL	
REACTOR VESSEL SUPPORT ANCHORS	
GARDNER ASSOCIATES, INC.	
APPROVED	
FLORIDA	
S-521-022, Rev 6	

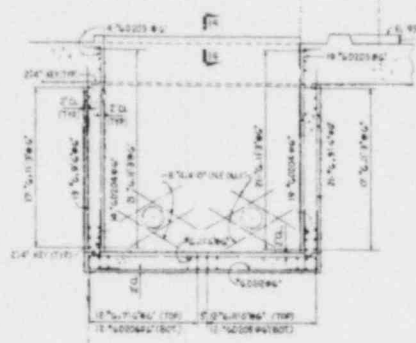




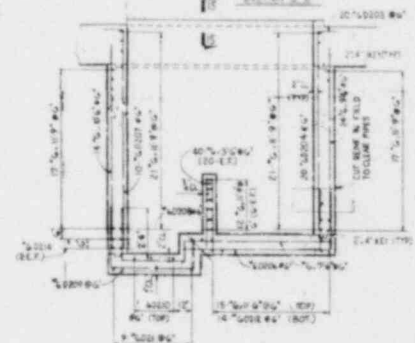
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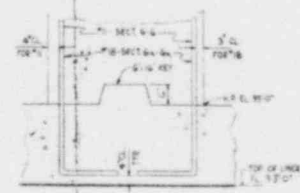
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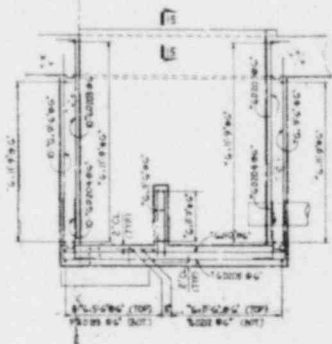
SECTION 2.2



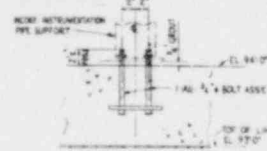
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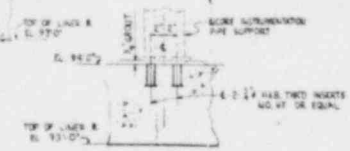
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SECTION 6.6a (SHOWN IN NOTES)



SECTION 5.5

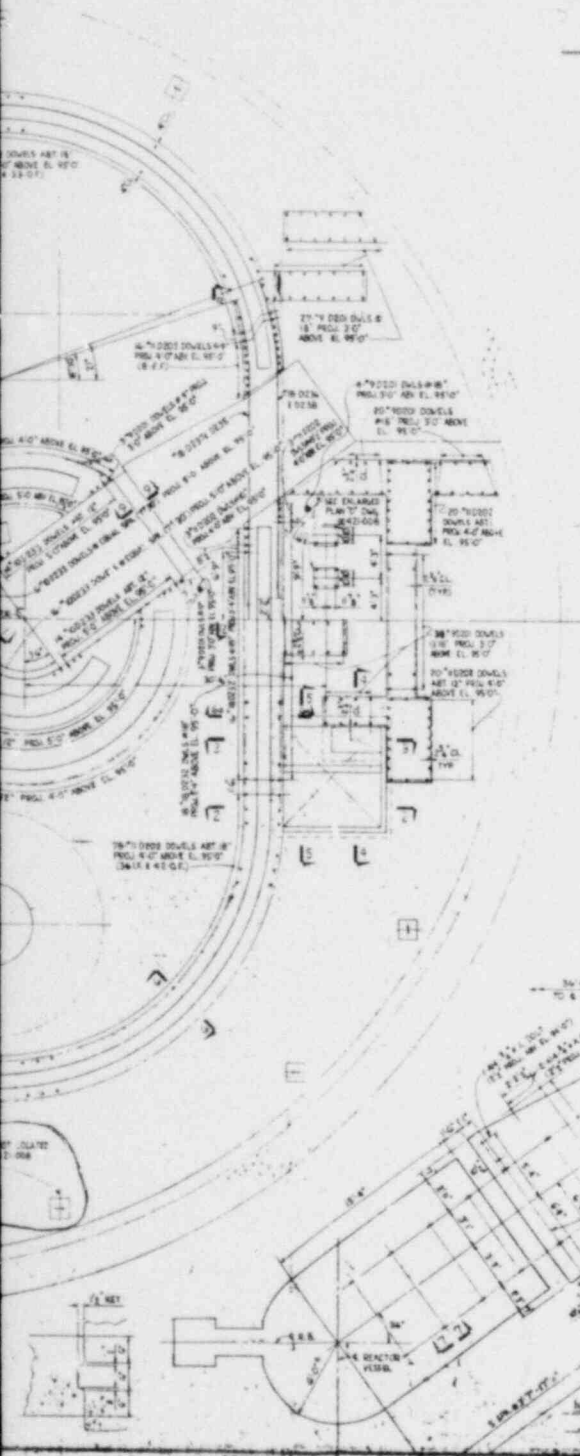


SECTION 7.7  
(TYP. IN PLACES)



SECTION 8.8  
(TYP. IN PLACES)

NOTES:  
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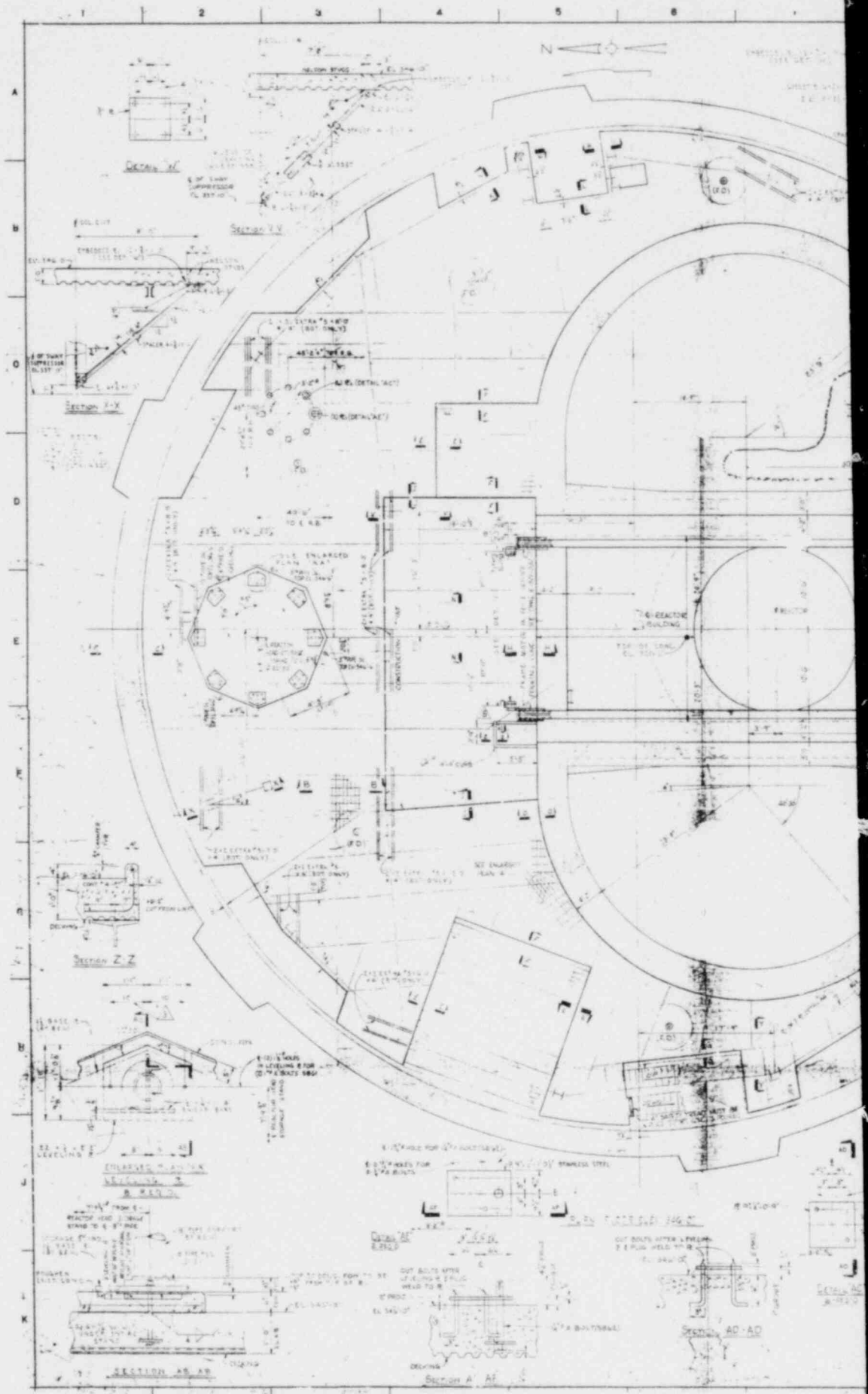
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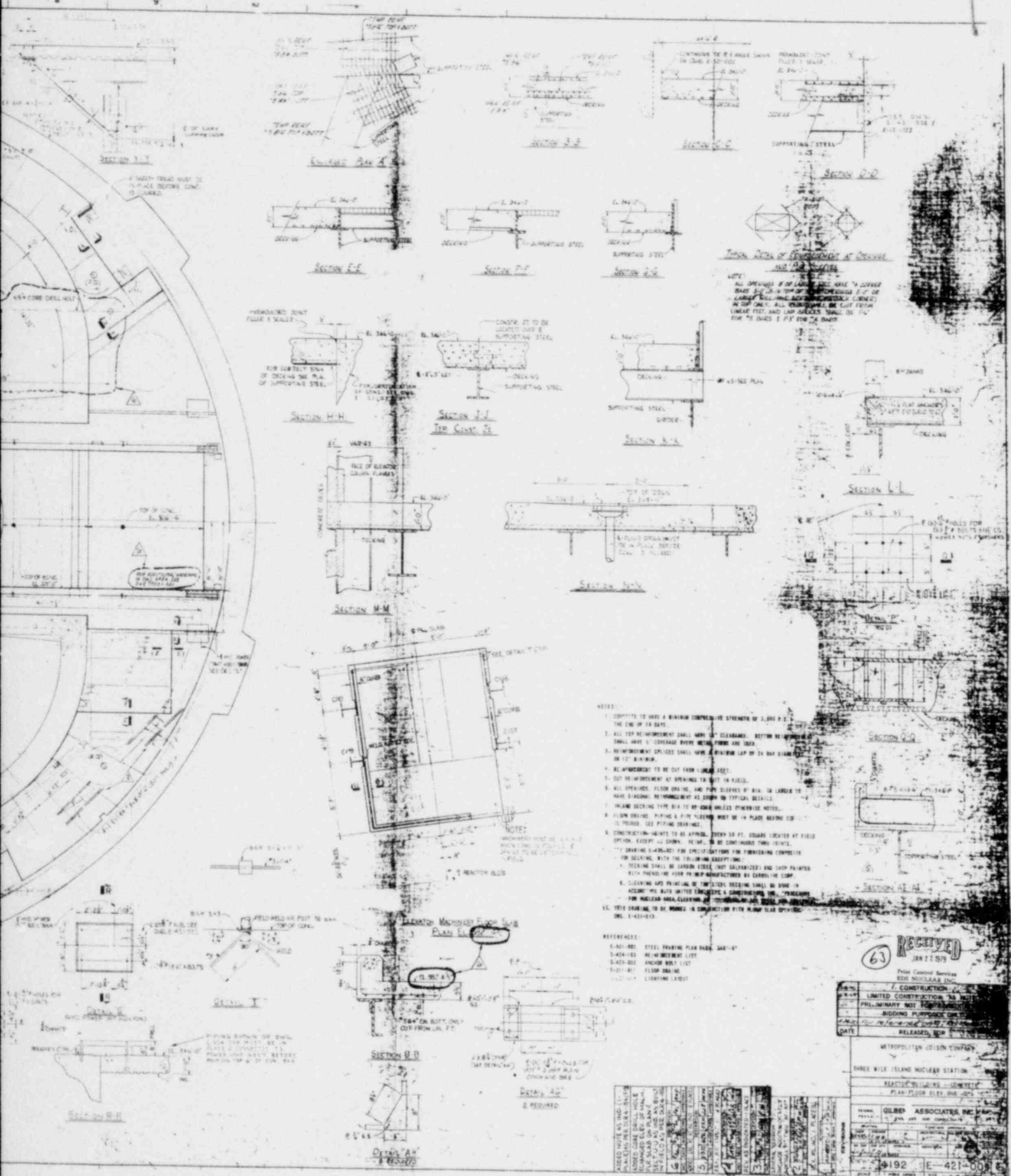
DESIGNED BY: W. A. BOLZ & ASSOC. S. BOLZ	DRAWN BY: J. L. HARRIS
<b>FLORIDA POWER CORPORATION</b>	
31 WASHINGTON, MIAMI	
CRYSTAL RIVER PLANT	
UNIT NO. 3	835,000 KW
REACTOR BUILDING BUILT & ANCHOR BUILT PLAN BASEMENT FLOOR E.L. 93'-0"	

CONTRACTOR: UNITED CONSTRUCTION CO. INC.	APPROVED FOR CONSTRUCTION: GILBERT ASSOCIATED, INC. ENGINEER AND ARCHITECT
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DATE: 12/20/56 SCALE: AS SHOWN SHEET NO.: 421-000-1	4800 18-481-000 SC-42-00
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- NOTES:
1. CONCRETE TO HAVE A MINIMUM COMPRESSIVE STRENGTH OF 3,000 P.S.I. THE THICKNESS OF SLABS SHALL BE 4" MINIMUM.
  2. ALL TOP REINFORCEMENT SHALL HAVE 180° BENDS. BOTTOM REINFORCEMENT SHALL HAVE 90° BENDS EXCEPT WHERE NOTED OTHERWISE.
  3. REINFORCEMENT SPACING SHALL HAVE A MINIMUM LAP OF 36 BAR DIAMETER OR 12" MINIMUM.
  4. REINFORCEMENT TO BE CUT FROM 4" MINIMUM.
  5. CUT REINFORCEMENT AT SPACING TO CUT IN FIELD.
  6. ALL OVERHEAD FLOOR SLABS AND POOR SLABS SHALL BE 4" MINIMUM TO 6" MAXIMUM. REINFORCEMENT TO BE 180° BENDS AT TOP AND 90° BENDS AT BOTTOM.
  7. FLOOR SLABS SHALL BE 4" MINIMUM TO 6" MAXIMUM. REINFORCEMENT TO BE 180° BENDS AT TOP AND 90° BENDS AT BOTTOM.
  8. FLOOR SLABS SHALL BE 4" MINIMUM TO 6" MAXIMUM. REINFORCEMENT TO BE 180° BENDS AT TOP AND 90° BENDS AT BOTTOM.
  9. CONSTRUCTION JOISTS TO BE APPROX. EVERY 30" ON CENTER LOCATED AT FLOOR SPACING EXCEPT AS NOTED. JOISTS TO BE CONTIGUOUS TWO SPANS.
  10. JOISTS TO BE 4" MAXIMUM FOR SPECIFICATIONS FOR FABRICATING COMPANIES FOR JOISTS. SEE THE FOLLOWING EXCEPTED:
  11. JOISTS SHALL BE CARBON STEEL, NOT GALVANIZED, AND SHALL BE PAINTED WITH FENOLIC WOOD PRESERVATIVE MANUFACTURED BY CAROLINE CORP.
  12. CLEARING AND FINISHING OF JOISTS TO BE DONE IN ACCORDANCE WITH THE BIDDING SPECIFICATIONS FOR JOISTS FOR NUCLEAR AREA. CLEARING OF JOISTS TO BE DONE IN ACCORDANCE WITH THE BIDDING SPECIFICATIONS FOR JOISTS FOR NUCLEAR AREA.
  13. JOISTS TO BE WELDED IN CONNECTION WITH FLOOR SLAB JOISTS. SEE 1-101-013.

- REFERENCES:
- 1-101-001 STEEL FRAMING PLAN DATA SHEET
  - 1-101-002 REINFORCEMENT LIST
  - 1-101-003 JOIST DATA SHEET
  - 1-101-004 FLOOR DATA SHEET
  - 1-101-005 JOISTING LIST

63

RECEIVED

JAN 27 1959

Point Control Section  
KBR NUCLEAR DIV.

CONSTRUCTION

LIMITED CONSTRUCTION TO BE  
PRELIMINARY NOT FOR BIDDING

ISSUED FOR INFORMATION ONLY

DATE RELEASED SEP 1958

METROPOLITAN EDISON COMPANY

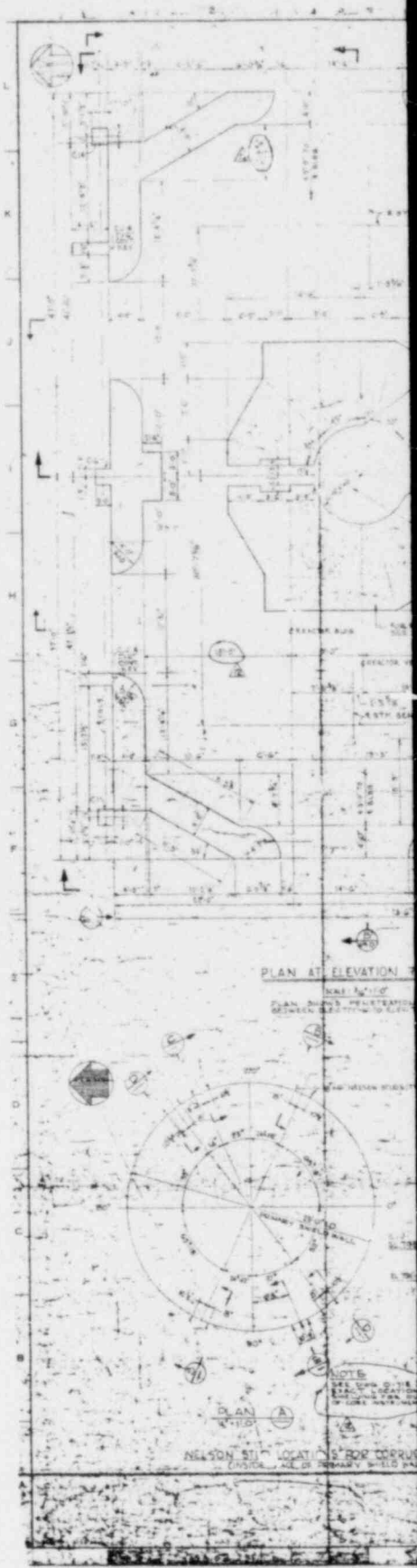
THREE MILE ISLAND NUCLEAR STATION

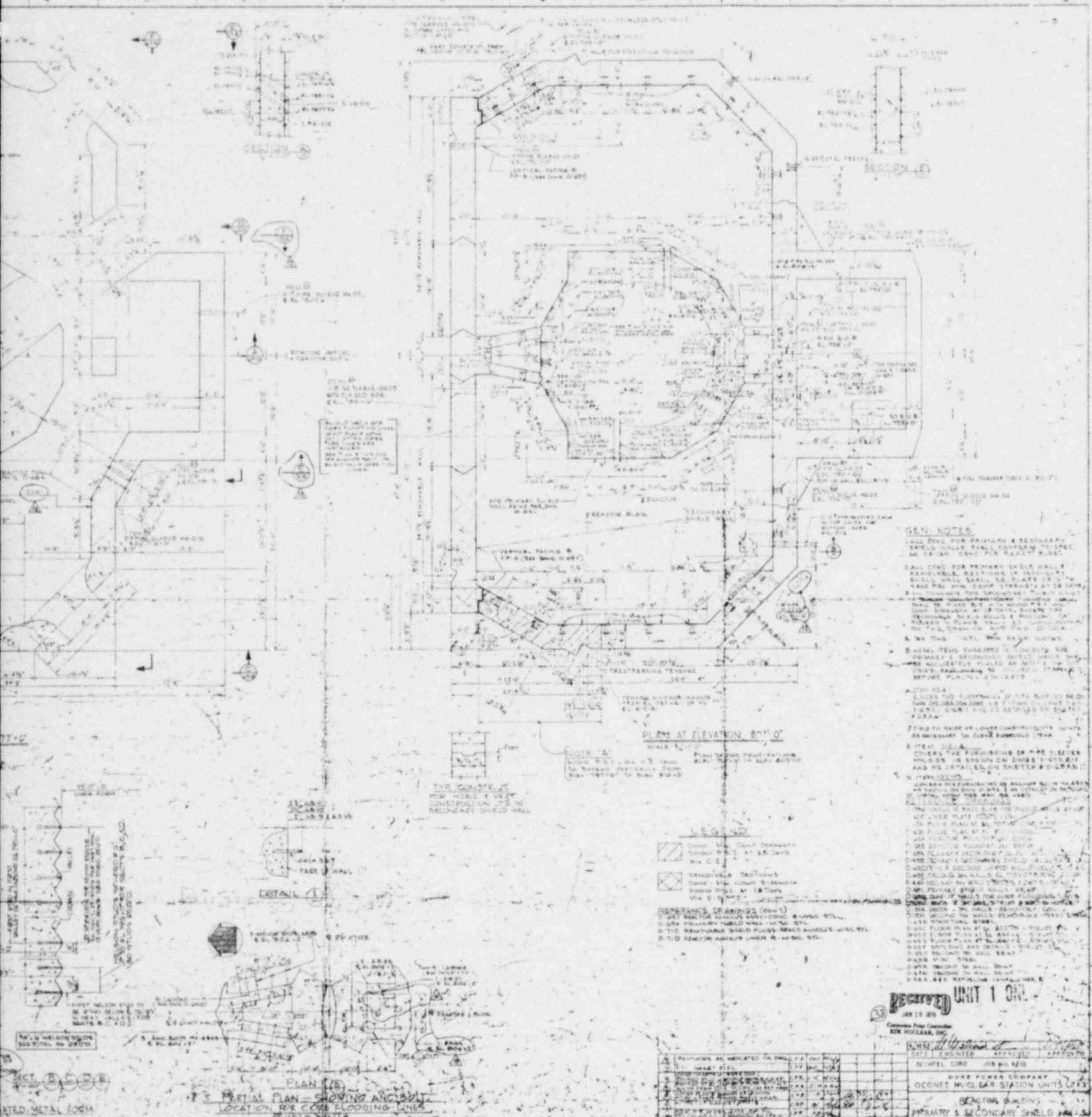
REACTOR BUILDING - LOWER LEVEL

PLAN-FLOOR ELEV. 800.000

GILBE ASSOCIATES INC. ENGINEERS

192 E-421-001





**PLAN AT ELEVATION 80' 0"**  
SCALE 1/8" = 1'-0"

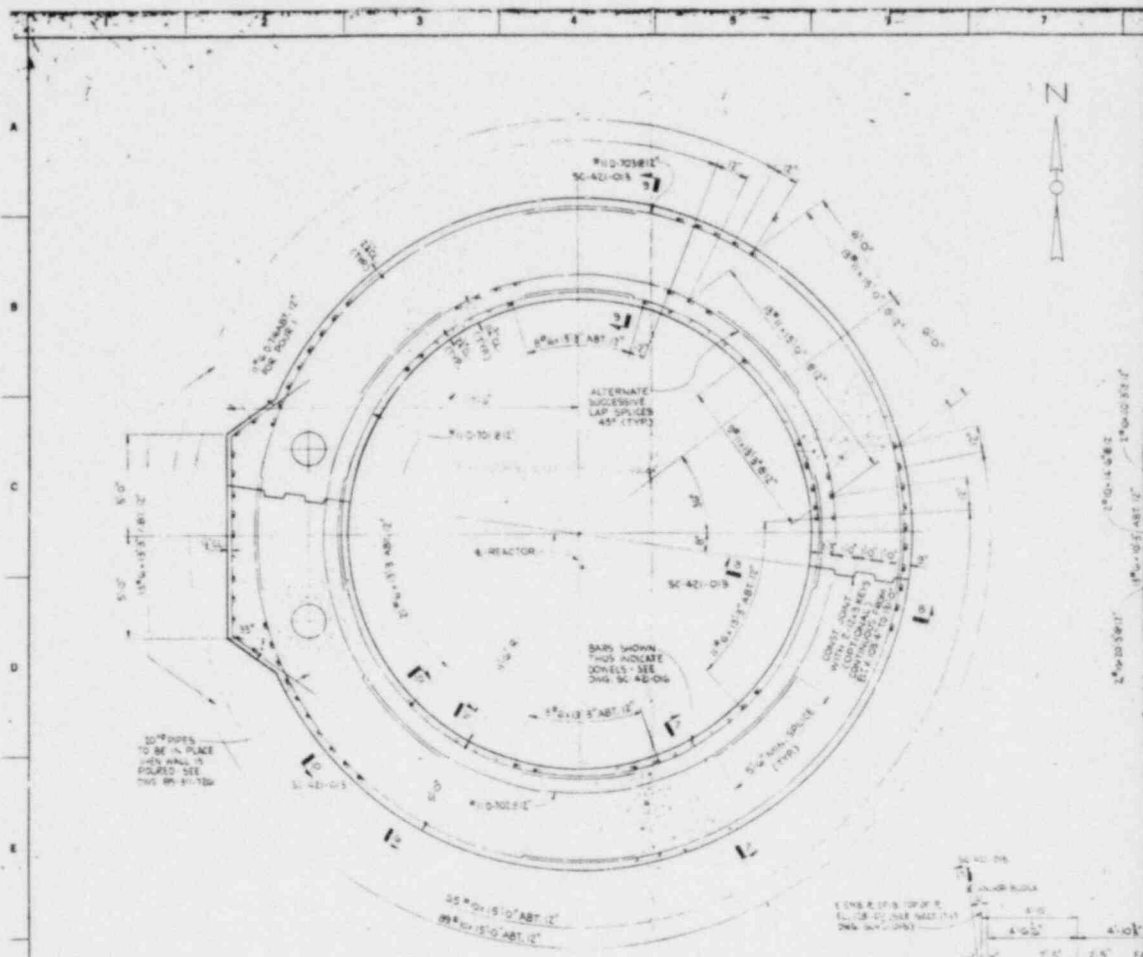
- LEGEND**
- CONCRETE SHIELD WALL  
CONCRETE SHIELD WALL AT 80' 0" ELEV.  
MA C. 2
  - REMOVABLE SECTIONS  
CONCRETE SHIELD WALL AT 80' 0" ELEV.  
MA C. 2
- REFERENCES DRAWINGS (Sheet)**
- D-207 SHIELD WALLS AND CORE WALLS
  - D-208 SHIELD WALLS AND CORE WALLS
  - D-210 SHIELD WALLS AND CORE WALLS

**GEN. NOTES**

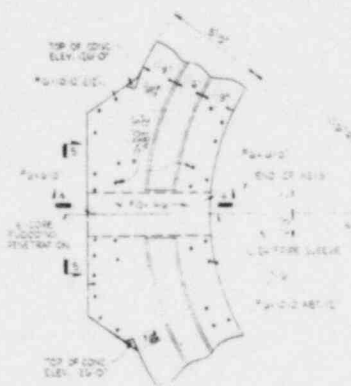
1. ALL WALLS FOR PRIMARY & SECONDARY SHIELD WALLS SHALL CONFORM TO SPEC. MA C. 2 OF THE CODE FOR SHIELD WALLS.
2. ALL WALLS FOR PRIMARY SHIELD WALLS & REMOVABLE SECTIONS OF SECONDARY SHIELD WALLS SHALL BE PLACED ON A 4" MIN. CONC. SLAB. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE.
3. ALL WALLS FOR SECONDARY SHIELD WALLS SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE. WALLS AT 80' 0" ELEV. SHALL BE CONCRETE.
4. ALL WALLS SHALL BE 18" THICK UNLESS OTHERWISE SPECIFIED.
5. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.
6. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.
7. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.
8. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.
9. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.
10. ALL WALLS SHALL BE FINISHED WITH INTERLOCKING CONCRETE BLOCKS.

**SECTION 1**  
PARTIAL PLAN - SHOWING ANCIENT LOCATION OF CORE FLOODING LINES  
SHIELD WALLS

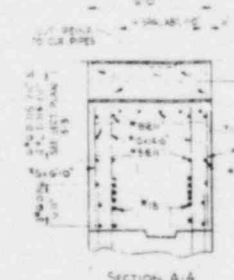
RECEIVED UNIT 1 031	
NO.	DESCRIPTION
1	CONCRETE SHIELD WALL
2	REMOVABLE SECTIONS
3	CONCRETE SHIELD WALL
4	CONCRETE SHIELD WALL
5	CONCRETE SHIELD WALL
6	CONCRETE SHIELD WALL
7	CONCRETE SHIELD WALL
8	CONCRETE SHIELD WALL
9	CONCRETE SHIELD WALL
10	CONCRETE SHIELD WALL
11	CONCRETE SHIELD WALL
12	CONCRETE SHIELD WALL
13	CONCRETE SHIELD WALL
14	CONCRETE SHIELD WALL
15	CONCRETE SHIELD WALL
16	CONCRETE SHIELD WALL
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18	CONCRETE SHIELD WALL
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21	CONCRETE SHIELD WALL
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29	CONCRETE SHIELD WALL
30	CONCRETE SHIELD WALL



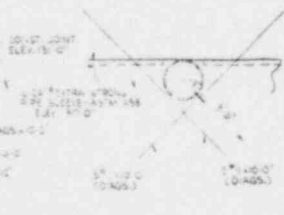
SECTIONAL PLAN 1-1



ELEVATION 13-0

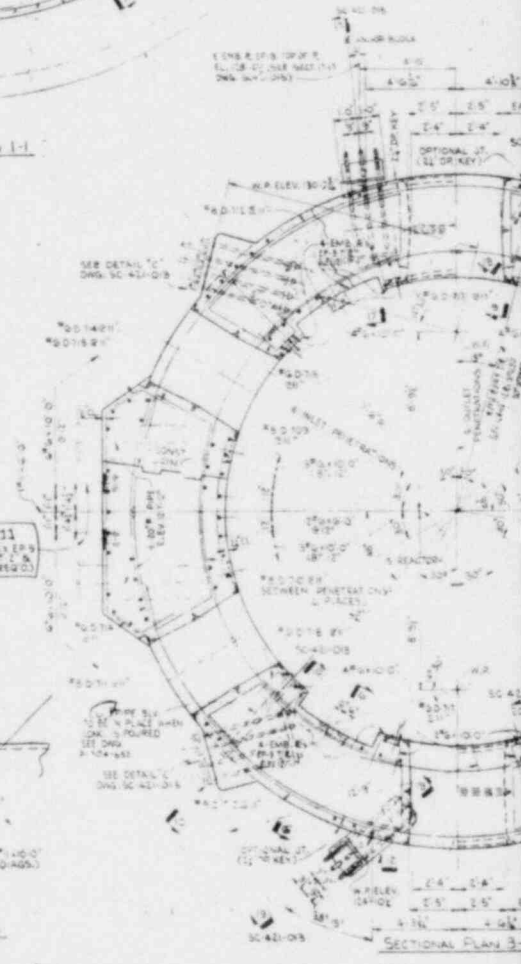


SECTION 4-4



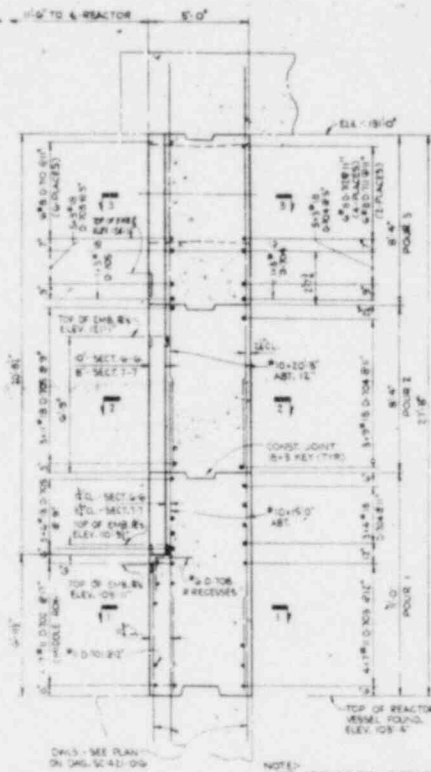
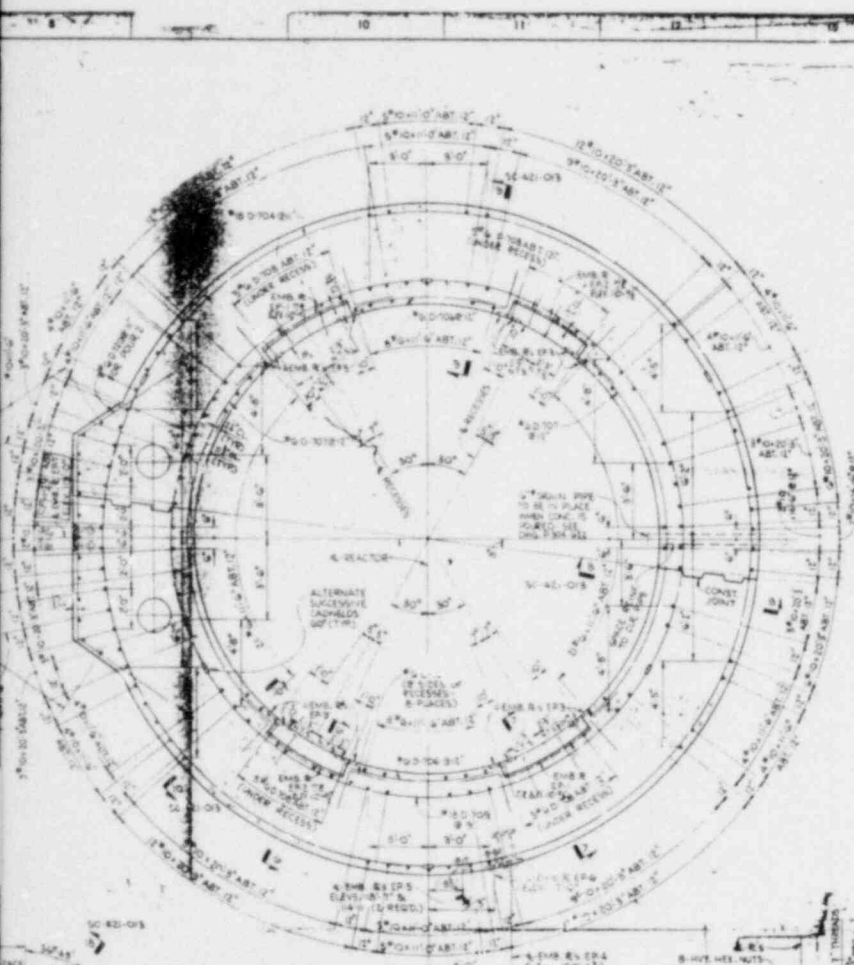
SECTION 5-5 (2 PLACES)

CP-11  
 1/2\"/>



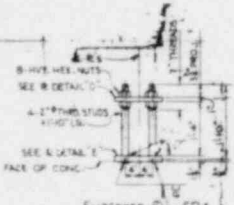
SECTIONAL PLAN 2-2



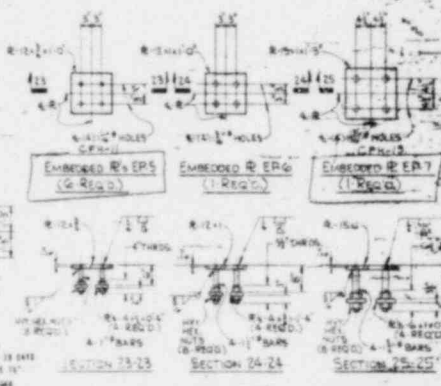


SECTION G-G  
SECTION 7-7 AS NOTED

SECTIONAL PLAN 2-2



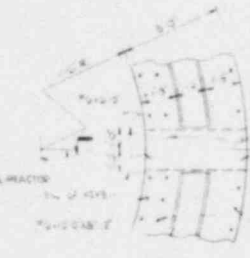
EMBEDDED R-EP4 (2 REQS)



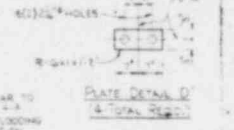
EMBEDDED R-EP4 (2 REQS)

EMBEDDED R-EP6 (1 REQ)

EMBEDDED R-EP7 (1 REQ)



PLAN Elevation 131'0"



EMBEDDED R-EP4 (2 REQS)

EMBEDDED R-EP6 (1 REQ)

SECTION 23-23

SECTION 24-24

SECTION 25-25

- NOTES:
- COMPLETE SPECIFICATIONS OF STEEL & OF STEEL STRENGTH & SIZE P.S. & NUMBER OF BARS SHALL BE - 5000-2 MAX. ADAPTIVE 1"
  - REINFORCEMENT SPECIFICATIONS OF STEEL & OF STEEL MATERIAL LIST A-513 SHALL BE AS
  - ALL REINFORCEMENT SHALL CONFORM TO SPECIFICATION FOR STRUCTURAL STEEL WITH AN-60T UNLESS OTHERWISE NOTED

- REFERENCES:
- SC-42-813 - REACTOR VESSEL STEEL SHIELD SHEET
  - SC-42-813 - REACTOR VESSEL STEEL SHIELD SHEET
  - SC-42-813 - REACTOR VESSEL STEEL SHIELD SHEET
  - SC-42-813 - REACTOR VESSEL STEEL SHIELD SHEET
  - SC-42-813 - REACTOR VESSEL STEEL SHIELD SHEET

RECEIVED  
APR 11 1968  
FLORIDA POWER CORPORATION  
ELECTRIC DIVISION  
CRYSTAL RIVER PLANT

CP-13-13

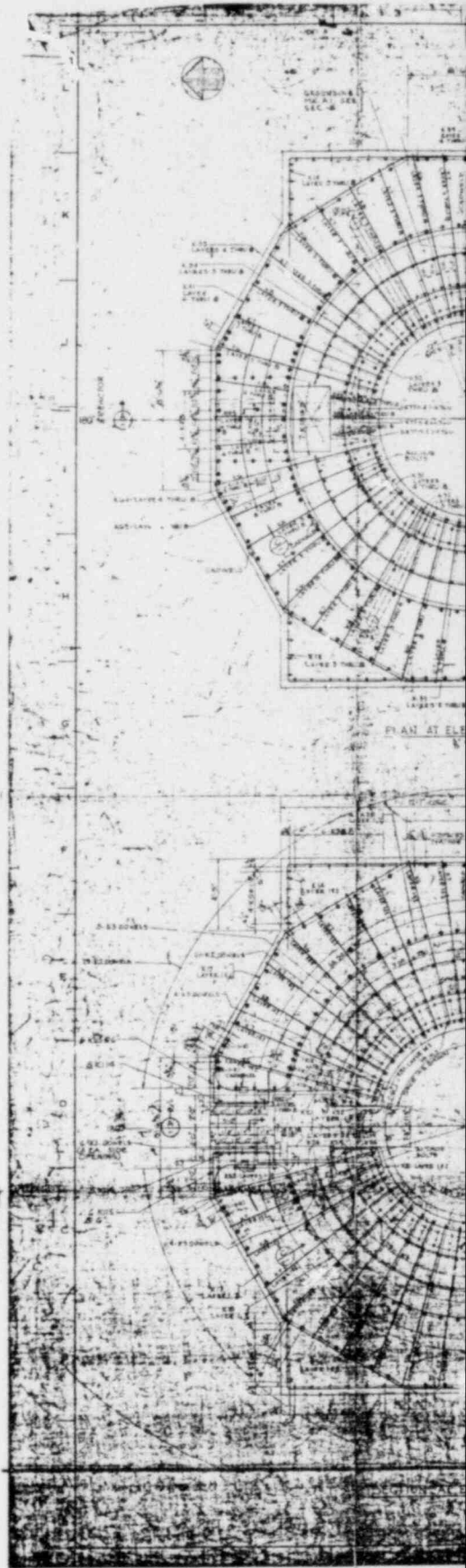
1	CHANGED SLEEVE SIZE SECT. PLAN 3-3	REVISION
2	ADDED REBAR EP4 & EP6 TO REACTOR VESSEL SHIELD SHEET	REVISION
3	ADDED REBAR EP4 & EP6 TO REACTOR VESSEL SHIELD SHEET	REVISION
4	ADDED REBAR EP4 & EP6 TO REACTOR VESSEL SHIELD SHEET	REVISION
5	ADDED REBAR EP4 & EP6 TO REACTOR VESSEL SHIELD SHEET	REVISION
6	ADDED REBAR EP4 & EP6 TO REACTOR VESSEL SHIELD SHEET	REVISION

**FLORIDA POWER CORPORATION**  
11 PETERSBURG ROAD  
CRYSTAL RIVER PLANT  
UNIT NO. 2 855,000 KW  
REACTOR VESSEL SHIELD SHEET  
PLATE & SECTIONS

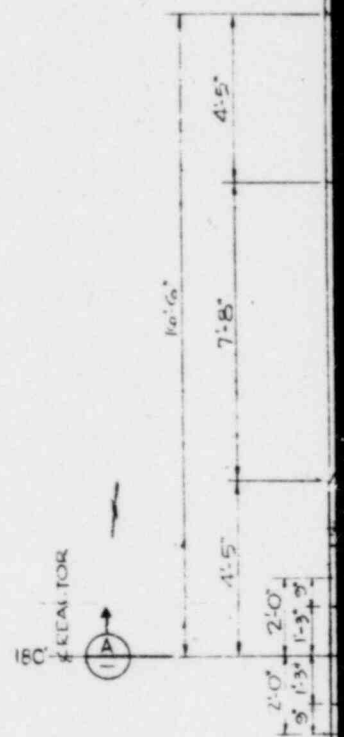
OLMSP ASSOCIATES, INC.  
1111 N. W. 11th St., Ft. Lauderdale, Fla.  
APPROVED FOR CONSTRUCTION  
DATE: 4/11/68  
BY: [Signature]

CONSTRUCTION AS NOTED  
LIMITED CONSTRUCTION AS NOTED  
PRELIMINARY NOT FOR CONSTRUCTION  
ISSUED FOR PURPOSES ONLY  
RELEASED FOR [Blank]

5 SC-42-813











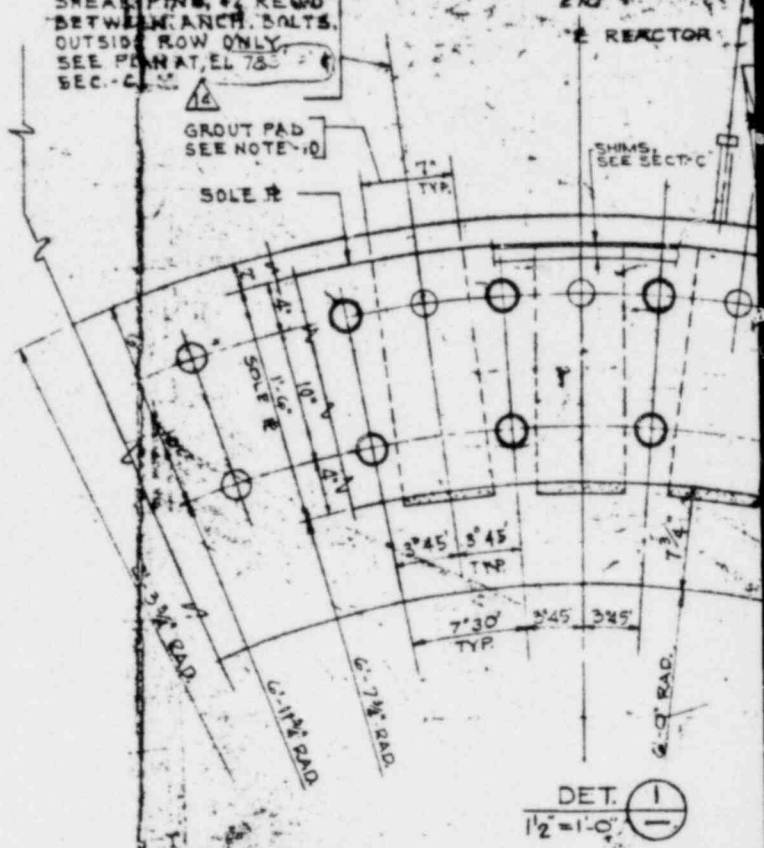
SHEAR PINS, 42 REQ'D  
BETWEEN ANCH. BOLTS,  
OUTSIDE ROW ONLY  
SEE PLAN AT EL 785'  
SEC. C

REACTOR

GROUT PAD  
SEE NOTE 10

SOLE PL

SHIMS  
SEE SECT C



DET. 1  
1/2" = 1'-0"

REACTOR BLDG.

5'-0" MIN 2'-0" 2'-4" 3'-0" 6'-0" RAD.

PRIMARY SHIELD WALL

OPEN

15'-7"

13'-11"

BOT OF REACTOR SKIRT PLANGE  
ELEV. 785'-6"

POUR CONCR  
ELEV. 785'-2"  
TO SET THEN  
ELEV. 785'-0"

ANCHOR BOLTS  
MK G1

CONTINUOUS KEY

FRAME ANCHOR  
DF-1

MK DF-4  
METAL DOOR  
DOOR FRAME  
MK DF-2

TOP CONCR  
EL VARIES

DET. 1  
FRAME ONLY  
0-480

BASEMENT FL

EL. 775'-0"

DET. 2  
0-520

TOP OF ANCHOR BOLT  
EL. 776'-1" (TYP)

BRICKED-UP ACCESS  
OPENING. USE SOLID  
CONC. BRICK WITH  
UNMORTARED JOINTS

REACTOR BLDG. BASE  
CLAD - SEE DWG. D-33E

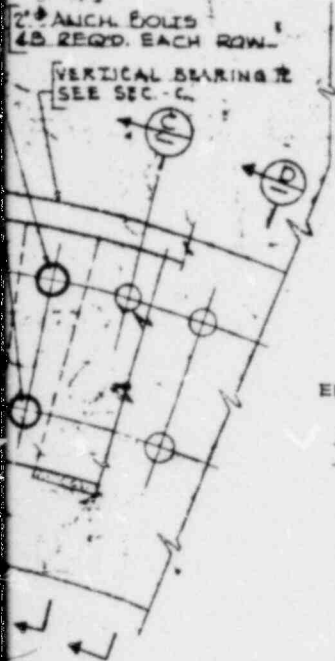
FASTEN LADDER  
TO CONCR WITH 1/2" +  
CLANCH ANCHORS  
SEE GEN. NOTE #6

SEC. A  
3/8" = 1'-0"

NELSON STUDS

2" ANCH. BOLTS  
(2 REQ'D. EACH ROW)

VERTICAL BEARING R  
SEE SEC. C



REACTOR SKIRT

JAM NUT  
2" HEAVY  
HEX. NUT  
2" STD. WASH  
HARDENED

ELEV. 786'+2"

WASHER  
2 1/2" ID. 4 1/8" O.D. 1" TH.  
MILL SOLE R  
EL. 785'+0"  
2" SOLE R  
5" GROUT SEE  
NOTE - 4

ELEV. 785'+0"

6'-0" RAD.  
6'-3 3/4" RAD.

4 1/4" RAD.

(SEE NOTE - 8  
SHEAR PIN 3/8" X  
IN 1 1/2" FIELD  
DRILLED HOLES  
(DRIVE FIT)

FILL WITH GROUT  
EL. 786'+0 1/2" HIGH PT. OF  
CONC. ALL AROUND SKIRT

LEAD IN GUIDE CAP  
3 REQ'D. TO BE PLACED  
OVER ALICAGE BOLTS P120.  
MK-GCF. SEE DWG.  
O-68C FOR DETAILS

CONST. JOINT

NELSON STUDS

VERTICAL BEARING R

2" ANCH. BOLTS

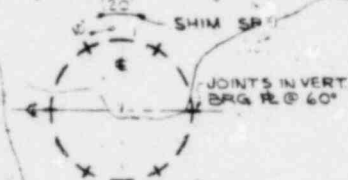
SEC. C  
1 1/2" - 1'-0"

IF THE SHIM(S) IS/ARE PLACED AT 2" AND  
REPEATED AT 2" INTERVALS, THE SHIMS  
WILL BE WITH JOINTS BETWEEN THE  
OF VERT. BRG. R.

SEC. D  
1 1/2" - 1'-0"

SPACE BETWEEN FLG. OF REACTOR  
SKIRT & SHIM (MK. SR5) TO BE  
FILLED WITH SHIMS FURNISHED  
IN FIELD AS REQUIRED.

NOTE:  
PLACE THIS CONC. AFTER  
PRIMARY SHIELD WALL  
& REACTOR HAVE BEEN  
INSTALLED.



PLAN UNIT-1243

12 SHIMS MK SR5, SPACED AT 30°

REACTOR

11'-6" RAD.

8'-3 3/4" RAD.

7'-9 3/4" RAD.

6'-11 3/4" RAD.

6'-0" RAD.

ELEV. 786'+2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

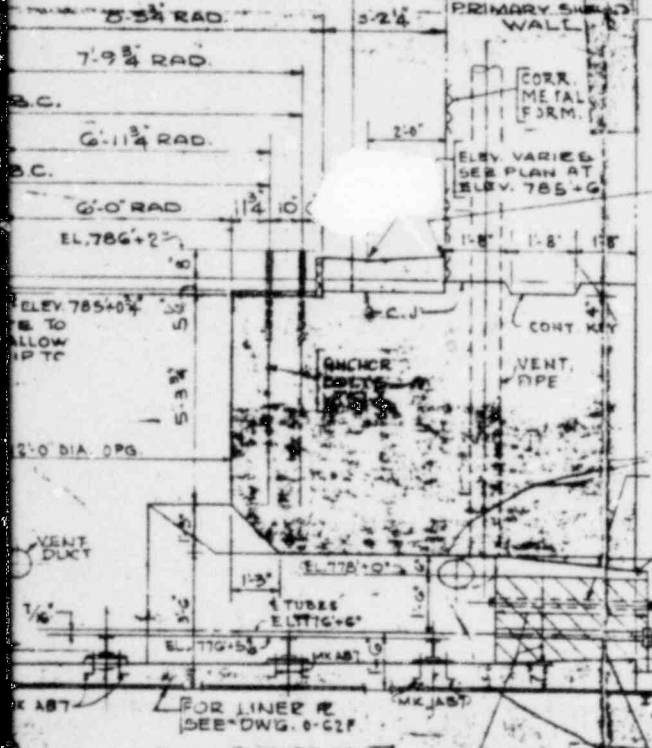
ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"

ELEV. 785'+0 1/2"



EL. 786'+0 1/2" HIGH PT. OF  
CONC. ALL AROUND  
PRIMARY SHIELD WALL

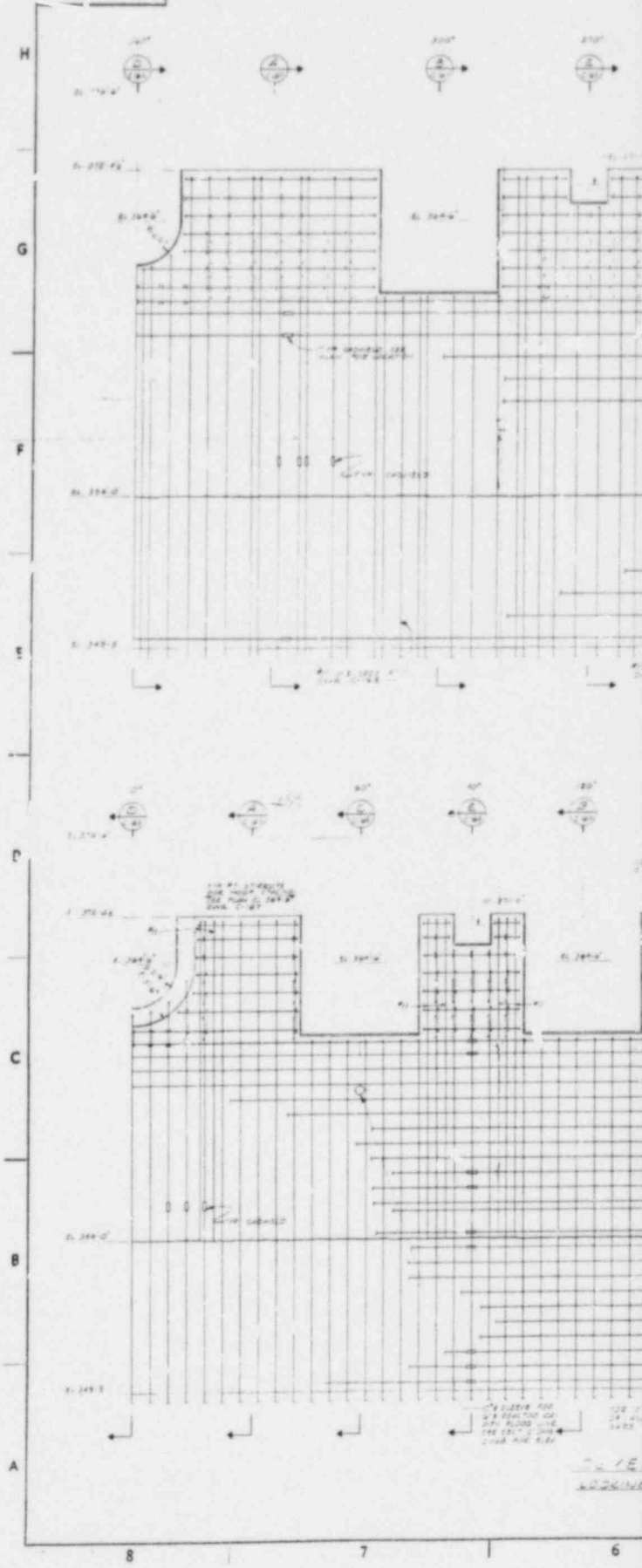
PRIMARY SHIELD  
WALL, SEE  
DWG. O-63A UNIT-1  
DWG. O-1069A UNITS 2&3

Dry Pack With Grout

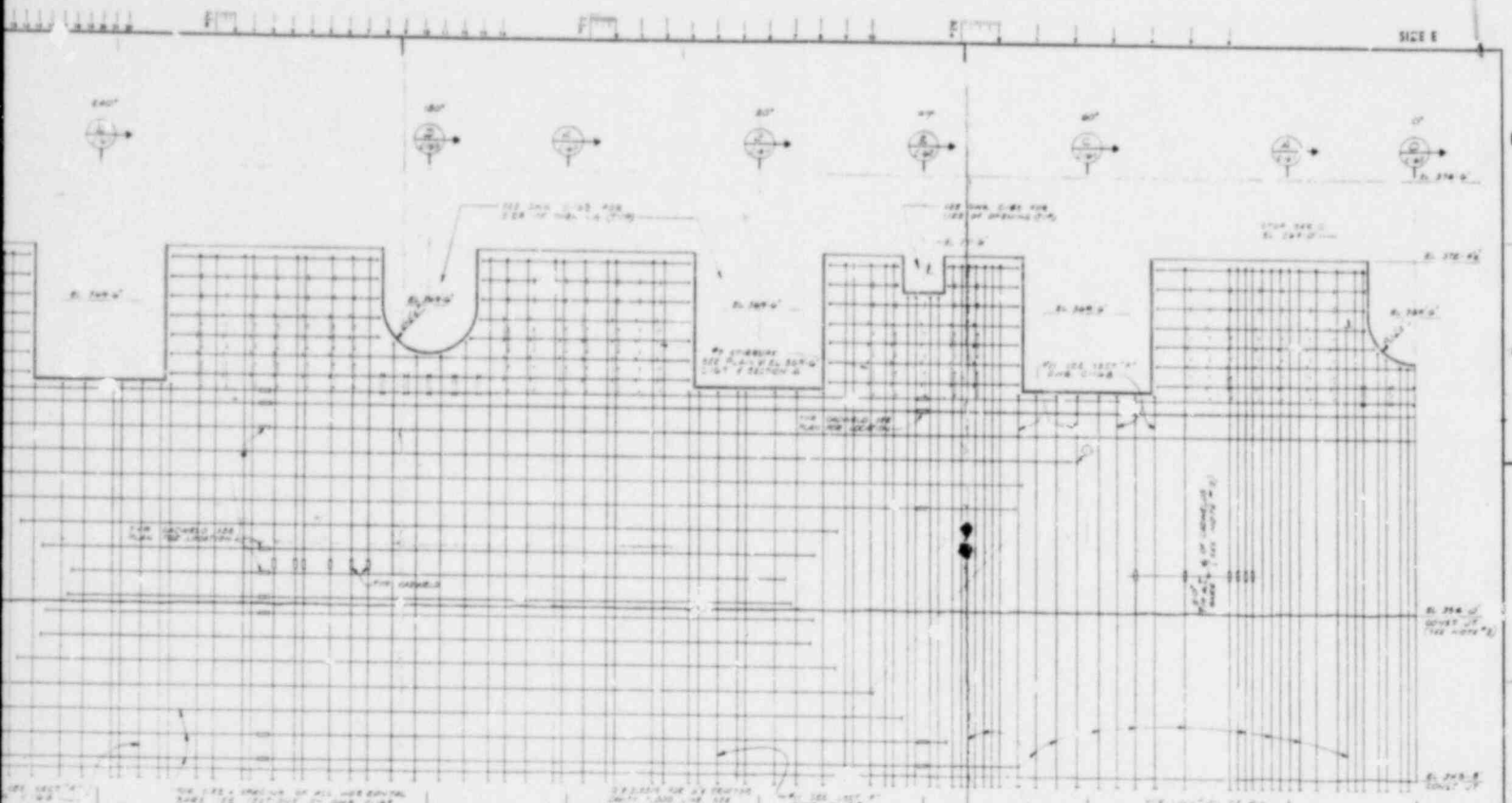
15

INCORE INSTRUMENTATION TUBE SLEEVES MKT51. SEE  
DWG. O-68C FOR DETAILS. SLEEVES TO BE ACCURATELY  
PLACED & SECURELY FASTENED BEFORE PLACING  
CLOSURE CONC. INCORE INST. GUIDE TUBES MK  
SUPPLIED BY B&W, MUST ALSO BE INSTALLED AND  
WELDED IN PLACE BEFORE PLACING CLOSURE CONC.

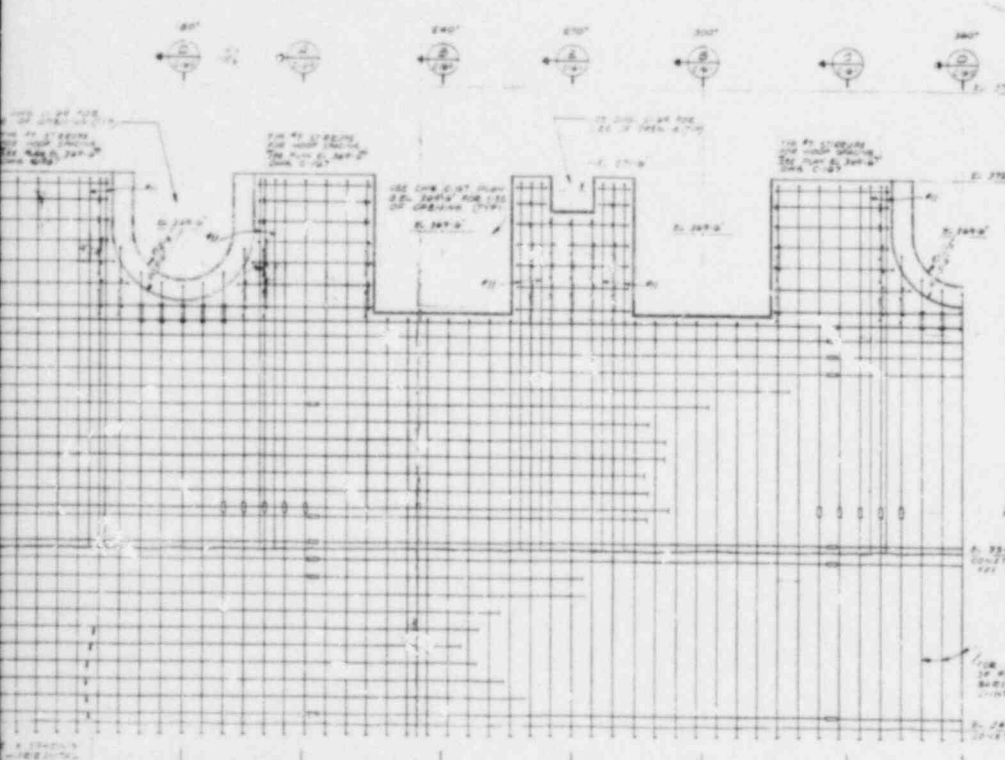
891-D-0000



891-D-0000



DEVELOPED ELEVATION 3 R-19-11  
LOOKING FROM OUTSIDE PRIMARY SHIELD WALL



DEVELOPED ELEVATION 3 R-19-11  
LOOKING FROM INSIDE PRIMARY SHIELD WALL

NOTES

1. SEE ADDITIONAL NOTES FOR THIS DRAWING
2. VERTICAL DIMENSIONS OF THIS DRAWING SHALL BE CONTROLLED BY CONTIGUOUS UNITS
3. CONSTRUCTION OF THIS ELEVATION SHALL BE CONTROLLED BY SECTION OF THE FIELD

Mr. J. W. Jones  
Mr. H. H. Hester  
Mr. D. A. Hester

INFORMATION ONLY

RECEIVED  
JAN 21 1975  
Plant Control Services  
KNS NUCLEAR INC.



1	2	3	4	5	6	7	8	9	10	11	12
REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS	REVISIONS
NO.	DATE	BY	DESCRIPTION	NO.	DATE	BY	DESCRIPTION	NO.	DATE	BY	DESCRIPTION

BECHTEL CORPORATION  
SAN FRANCISCO

ARKANSAS POWER & LIGHT COMPANY  
ARKANSAS NUCLEAR ONE

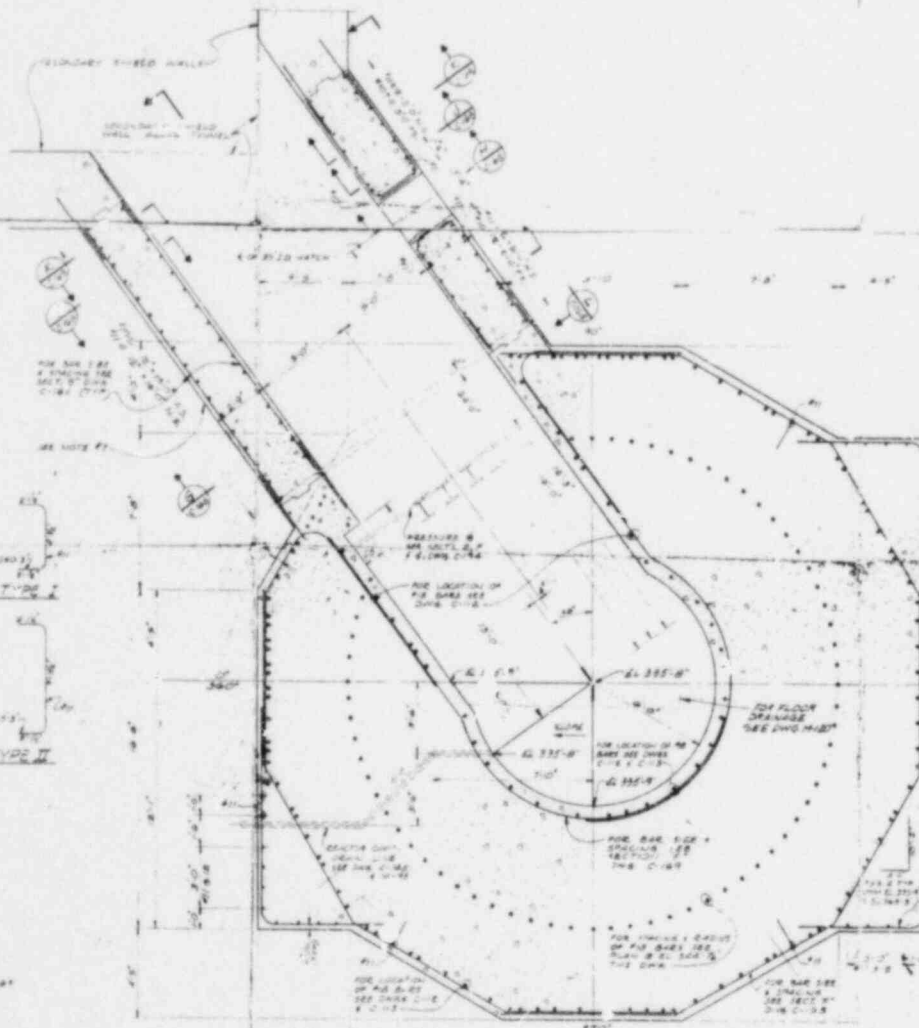
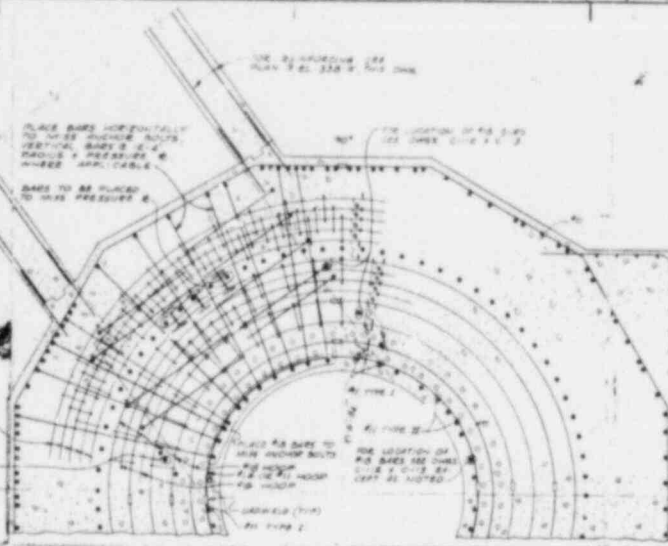
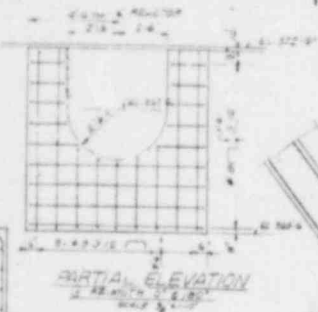
REACTOR BUILDING  
PRIMARY SHIELD  
DEVELOPED ELEVATION

6600 C-169



NOTES

1. THE GENERAL NOTES ARE ON SHEET C-100
2. ALL #4, #6 & #8 BARS SHALL BE PLACED WITH THE PRIMARY SHIELD WALL, UNLESS OTHERWISE NOTED TO BE PLACED AT THE SECONDARY SHIELD WALL. ALL OTHER BARS SHALL BE PLACED UNLESS OTHERWISE NOTED TO BE PLACED AT THE SECONDARY SHIELD WALL.
3. ALL JOISTS SHALL BE SPACED ACCORDING TO THE SPECIFICATION AND SHALL BE SPACED UNLESS OTHERWISE NOTED.
4. ALL JOISTS TO BE SPACED ACCORDING TO THE SPECIFICATION UNLESS OTHERWISE NOTED.
5. MINIMUM LENGTHS OF HORIZONTAL REINFORCING BARS:
  - #4 BARS 40'-0"
  - #6 BARS 40'-0"
  - #8 BARS 40'-0"
  - #10 BARS 40'-0"
  - #12 BARS 40'-0"
6. MINIMUM ENCHANCEMENT OF EMBEDMENT LENGTHS TO BARS 40'-0"
7. IN CASE OF PRECAST CONCRETE TUBES, CONCRETE SHALL NOT BE PLACED UNTIL AFTER INSTALLATION OF THE IN-CORE INSTRUMENTATION TUBING.
8. PRIMARY SHIELD WALL SHALL BE 8000 PSI.
9. SECONDARY SHIELD WALL SHALL BE 8000 PSI.



REFERENCE DWGS

- C-100 PLUMB PLANS & DETAILS SHEET 1
- C-100 PLUMB PLANS & DETAILS SHEET 2
- C-100 PLUMB PLANS & DETAILS SHEET 3
- C-100 DEVELOPED ELEVATION
- C-100 FOR PRECAST CONCRETE TUBING & SUPPORTS
- C-100 PRIMARY SHIELD WALL TYPE & PLAN
- C-100 SECONDARY SHIELD WALL TYPE & PLAN
- C-100 REINFORCING BARS & DETAILS W/EL. DETAILS
- C-100 PLAN & ELEVATION TO 340'-0"

RECEIVED  
JAN 31 1957

INFORMATION ONLY

REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY
REACTOR BUILDING	REINFORCING PLANS	DATE	BY

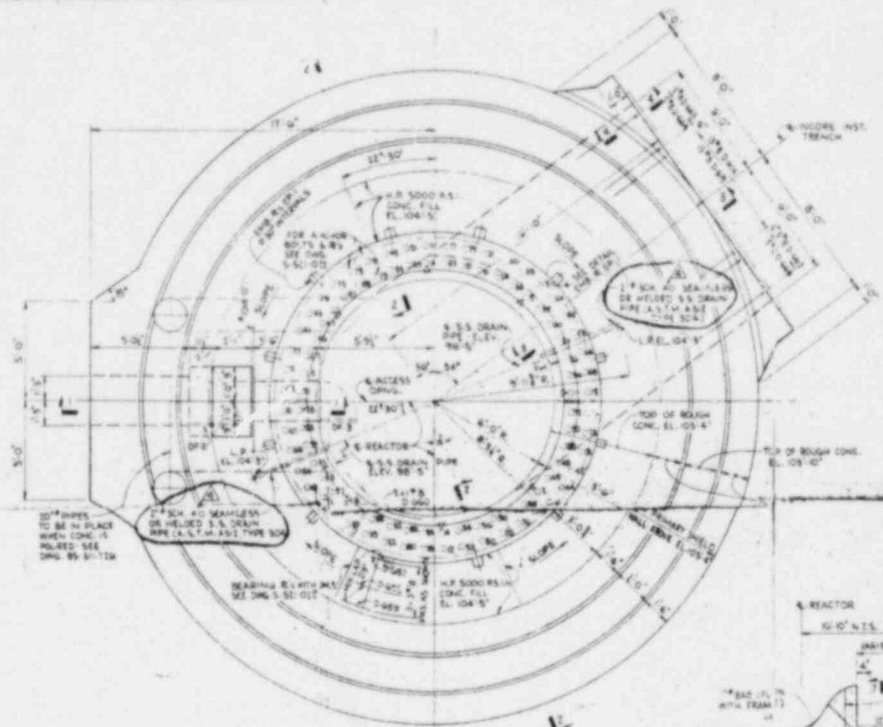
**BECHTEL CORPORATION**  
SAN FRANCISCO

ARKANSAS POWER & LIGHT COMPANY  
ARKANSAS NUCLEAR ONE

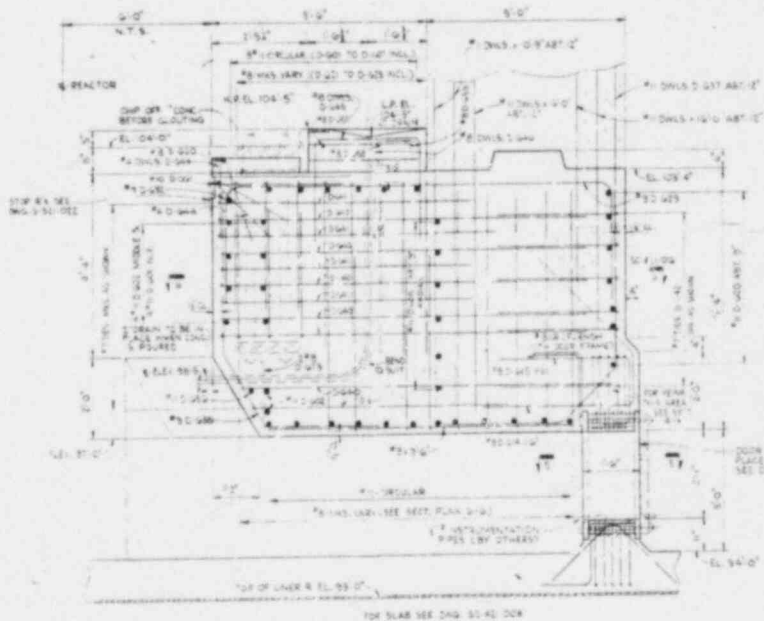
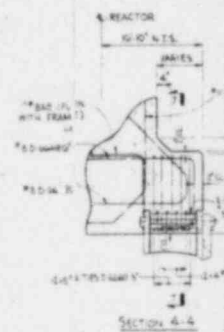
REACTOR BUILDING  
PRIMARY SHIELD  
REINFORCING PLANS

6600 C-167 16

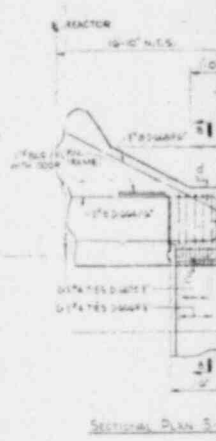




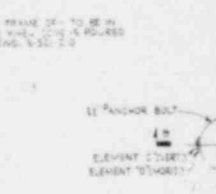
PLAN  
REACTOR VESSEL FOUND.



SECTION 3-3

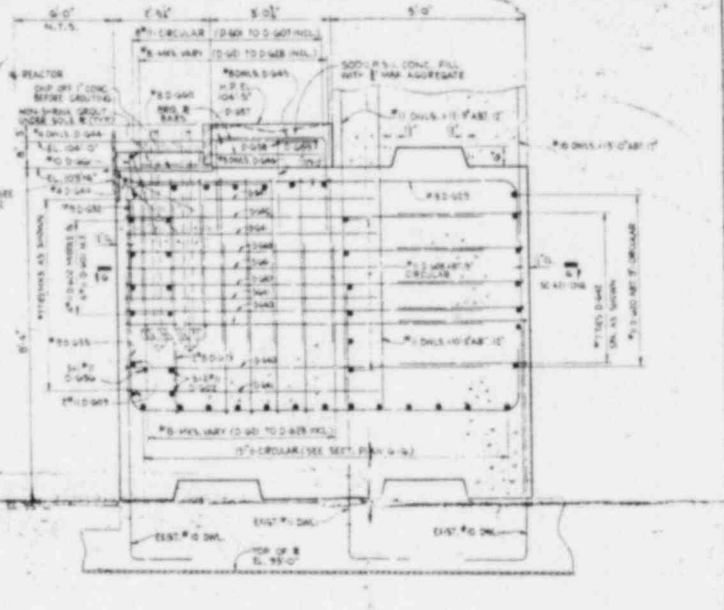
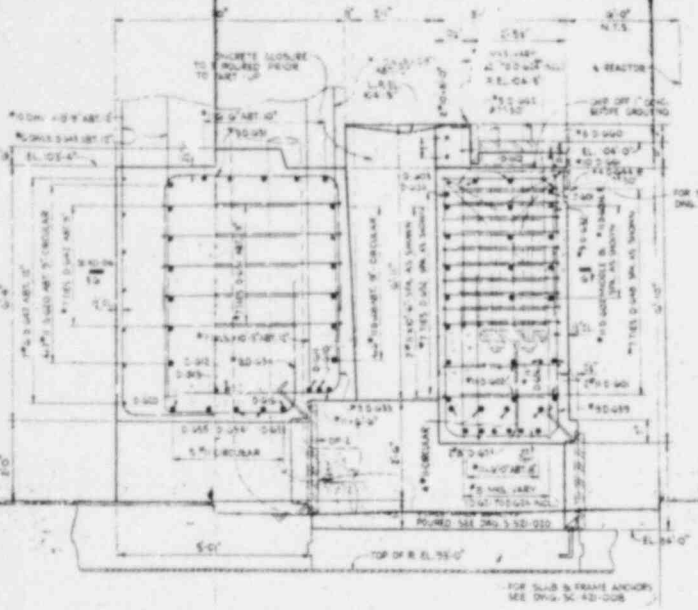


SECTIONAL PLAN 3-3



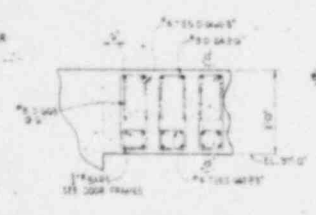
ANCHOR BOLT  
ELEVANT STUDS  
ELEVANT STUDS





SECTION 1-1

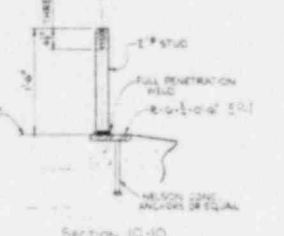
SECTION 2-2



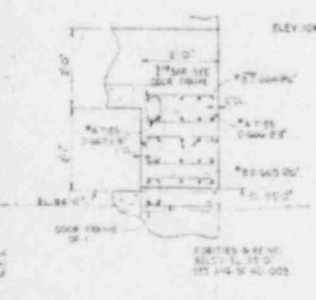
SECTION 7-7



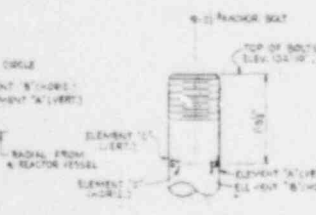
DETAIL EMBEDDED PLATE (EPI) FOR INSERVICE INSPECTION GEAR (I.I. REQUIRED)



SECTION 10-10



SECTION 5-5



SECTION 4-9

- NOTES
- CONCRETE SPECIFICATIONS OF THIS AND OTHER STRENGTH 4,000 P.S.I. MINIMUM AT 28 DAYS. CLASS TOP-3000-1 SHALL AGGREGATE 1 1/2\"/>
  - REINFORCING SPECIFICATIONS OF THIS AND OTHER STRENGTH 4,000 P.S.I. MINIMUM AT 28 DAYS. CLASS TOP-3000-1 SHALL AGGREGATE 1 1/2\"/>
  - REINFORCING SHALL BE NON-METALLIC, DIMENSIONALLY STABLE, PROTECTED AND NON-BRITTLE. 'STAINLESS STEEL' SHALL BE MANUFACTURED BY CARBORUNDUM CORP. OR BY FURNACE ALL IN ACCORDANCE WITH THE MANUFACTURER'S INSTRUCTIONS. THE NON-METALLIC REINFORCING SHALL BE A MINIMUM OF 1/4\"/>
  - SECTION 10-10: EMBEDDED STRAIN GAUGES SHALL BE CEMENTED TO THE CONCRETE AND LOCATED ON THE UPPER SURFACE AS SHOWN ON 'REACTOR VESSEL FOUNDATION'. THE STRAIN GAUGES SHALL BE PROTECTED FROM DAMAGE BEFORE THE SURFACE CONCRETE IS PLACED. A WATERPROOF AND SILENT FLOOR FINISH FOR THE YEARS AFTER INITIAL SERVICE OPERATIONS OF THE REACTOR VESSEL SHALL BE THE STRAIN GAUGES SHALL BE 'TRA-4 TYPE (PART-250-200)' AS MANUFACTURED BY CALDWELL-INDEPENDENT. THE STRAIN GAUGES AND ASSOCIATED INSTRUMENTATION SHALL HAVE A RESOLUTION OF 100 (1) MICRO INCHES PER INCH. STRAIN GAUGES SHALL BE FURNISHED, DETAILING AND SHOP DRAWING BY AN INDEPENDENT TESTING LABORATORY. GAUGES TO BE USED IN THIS TYPE OF WORK. THE TESTING LABORATORY SHALL SUBMIT INSTRUMENTATION DETAILS AND DRAWINGS TO THE ENGINEER PRIOR TO INSTALLATION OF THE STRAIN GAUGES. THE STRAIN GAUGES OF THE STRAIN GAUGES TO THE BOLTS SHALL ACCORDING TO A BUREAU ARTICLE. SHOP DRAWING TO THE MODEL STRAIN GAUGES SHALL BE MADE IN A 100 (1) ACTIVE AND STRAIN GAUGES INSTEAD OF 100 (1) INSTRUMENTATION EQUIPMENT SHALL BE CAPABLE OF MEASURING APPROXIMATELY 100 (1) MICRO INCHES PER INCH. THE STRAIN GAUGES SHALL BE MONITORED AT THE TIME OF INSTALLATION AND AFTER INSTALLATION. NO CHANGES TO THE GAUGES, CALIBRATION AND STRAIN DATA SHALL BE SUBMITTED TO THE ENGINEER. FOR PROTECTING FROM DAMAGE TO THE STRAIN GAUGES AND INSTRUMENTATION, 'WATERPROOF AND SILENT FLOOR FINISH' SHALL BE PLACED ON 'REACTOR VESSEL FOUNDATION' TO THE GRADE OF THE TOP SURFACE OF THE REACTOR VESSEL.

- REFERENCES:
- SC-42-015-1 REACTOR VESSEL FOUNDATION REINFORCEMENT PLACING
  - SC-42-015-2 CONCRETE POURING AND CURE PLAN
  - SC-42-015-3 WELD PLATE & BRACKET DRAWING
  - SC-42-015-4 REINFORCEMENT LIST
  - SC-42-015-5 REINFORCEMENT LIST
  - SC-42-015-6 REINFORCEMENT LIST
  - SC-42-015-7 REINFORCEMENT LIST
  - SC-42-015-8 REINFORCEMENT LIST

RECEIVED

DATE

FILE

FLORIDA POWER CORPORATION  
 ST. PETERSBURG, FLORIDA  
 CRYSTAL RIVER PLANT  
 UNIT NO. 3 825,000 KW

REACTOR BUILDING  
 REACTOR VESSEL FOUNDATION  
 PLAN & SECTIONS

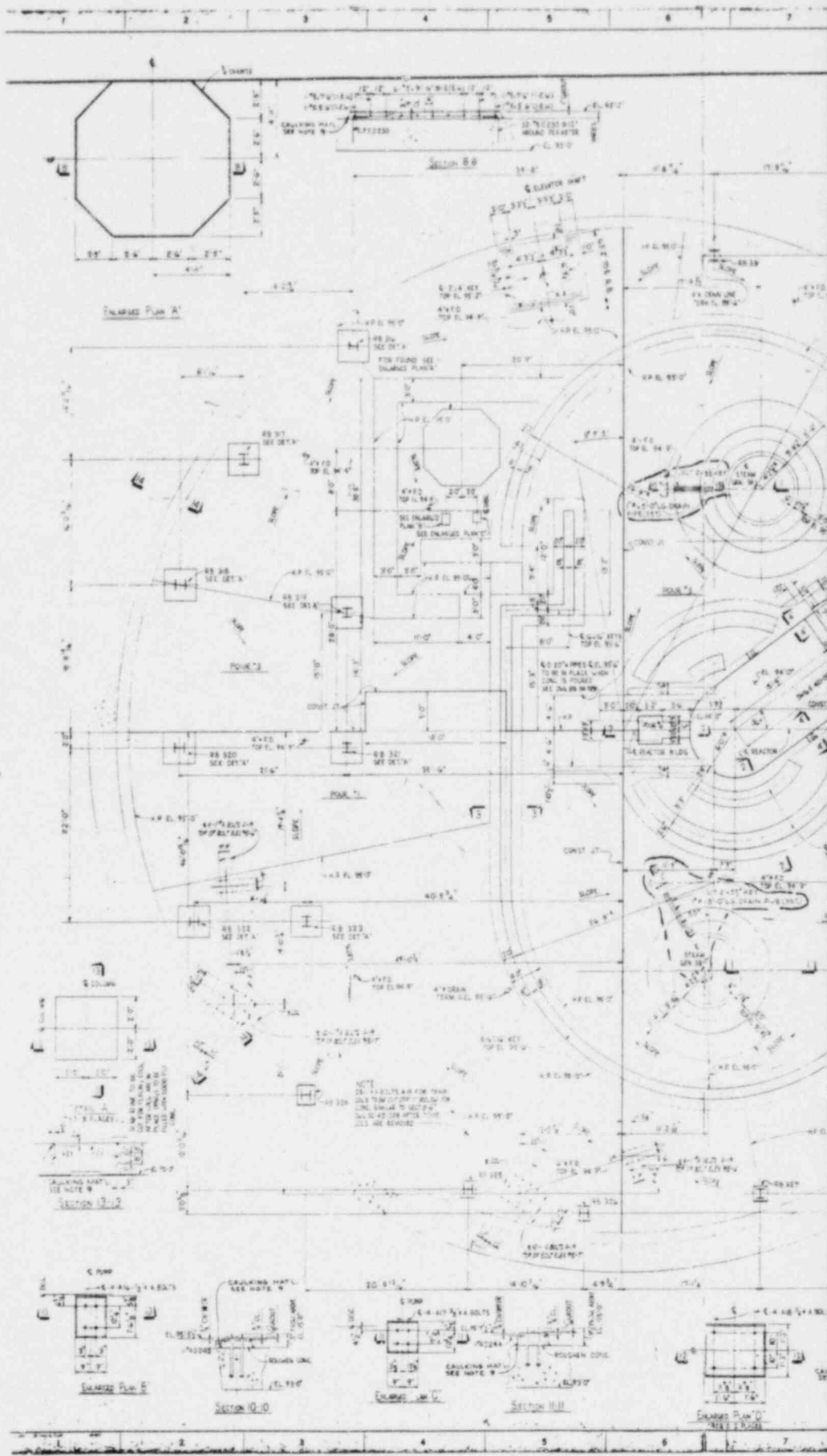
APPROVED FOR CONSTRUCTION BY

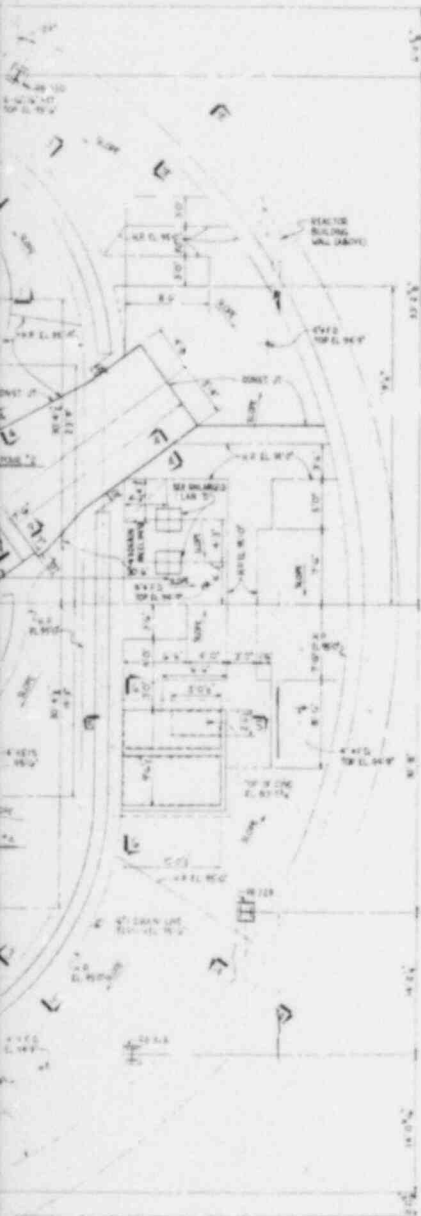
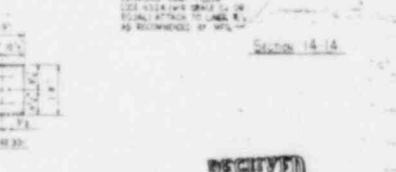
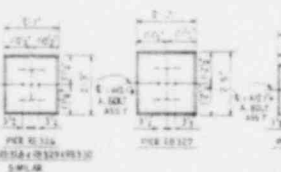
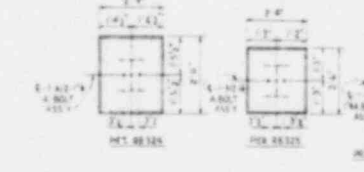
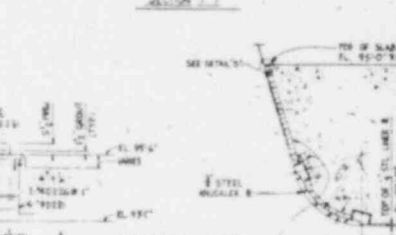
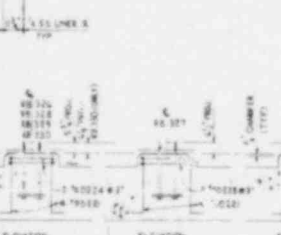
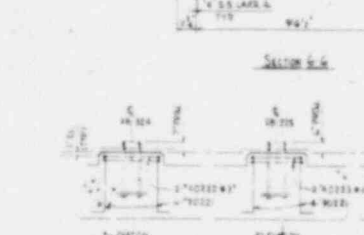
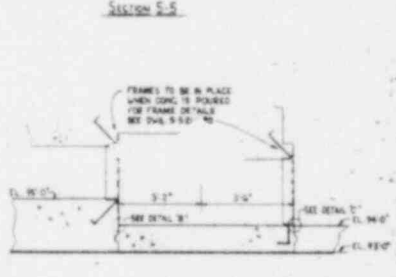
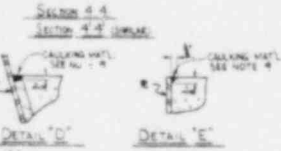
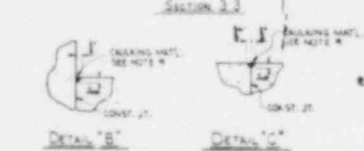
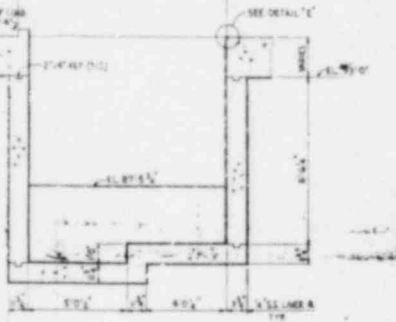
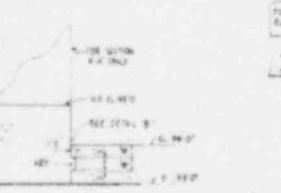
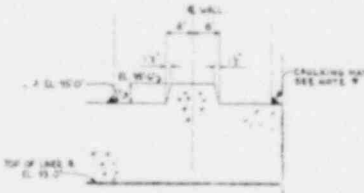
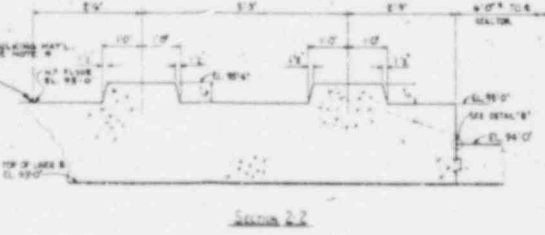
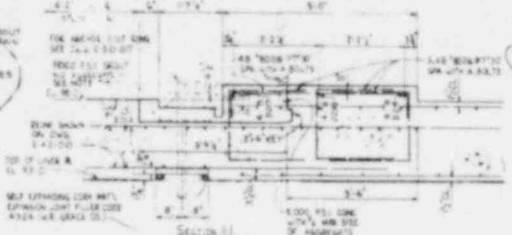
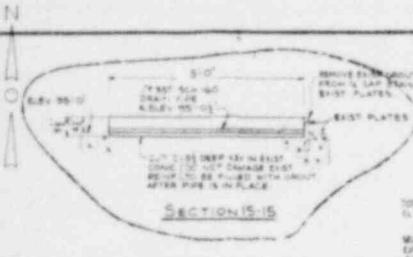
SC-42-015

DATE	RELEASED FOR	ENGR	SC-42-015	4	SC-42-015
DATE	RELEASED FOR	ENGR	SC-42-015	4	SC-42-015

NOTES

- FOR LOCATION OF STRAIN GAUGES SEE BOLTS MARKED 'TOP OF BOLTS' ON PLAN REACTOR VESSEL TO BE
- INSTALL STRAIN GAUGES AT 1/2\"/>
- CONCRETE HAS BEEN CAPPED AND READY TO RECEIVE BOLT PLATE.





- NOTES:**
1. CONCRETE SPECIFICATIONS - SP-4085 & SP-4011. FORMS TO BE USED PER 8-10-68 AT 28 DAYS. CURED BY 70°F. & 50% HUMIDITY MIN.
  2. REINFORCING SPECIFICATIONS SP-1004 & SP-1014. MATERIAL - ASTM A615 GR. 60.
  3. CONSTRUCTION DETAILS FOR WALLS & COLUMN BEAMS. SEE NOTES AND DRAWING SHEET OF BEAMS.
  4. FLOOR BEAMS & PILING TO BE IN PLACE BEFORE CONCRETE IS POURED.
  5. ALL CONSTRUCTION DETAILS & DIMENSIONS OF CONCRETE PLANS. SHALL HAVE CHANGED DIMENSIONS SHOWN IN DETAILS. SEE THIS DRAWING.
  6. FLOOR BEAMS TO RECEIVE A 2\"/>

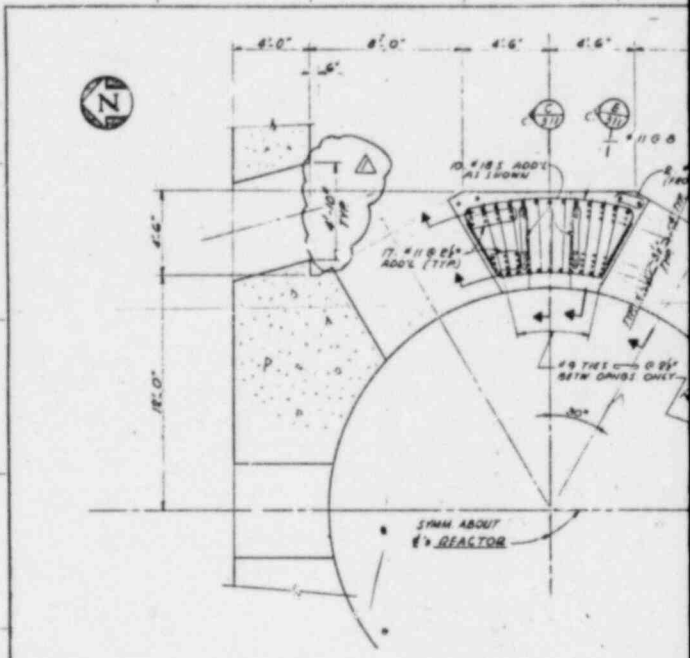
- REFERENCES:**
- SC-421-008 BOMB & MOUND BOLT PLAN
  - SC-421-018 REINFORCING PLAN
  - SC-421-023 FLOOR BEAM LIST
  - SC-421-042 FLOOR BEAM LIST
  - SC-421-043 MOUND BOLT LIST
  - SC-421-053 REINFORCING LIST
  - SC-421-063 REINFORCING LIST
  - SC-421-064 REINFORCING LIST
  - SC-421-065 REINFORCING LIST
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  - SC-421-098 REINFORCING LIST
  - SC-421-099 REINFORCING LIST
  - SC-421-100 REINFORCING LIST

**RECEIVED**  
APR 14 1967

FLORIDA POWER CORPORATION  
 4205 E-421-008-5  
 UNIT NO. 3 233,000 KW  
 REACTOR BUILDING  
 CONCRETE OUTLINE  
 BASEMENT FLOOR EL. 93.0'

**APPROVED**  
 GIBNEY ASSOCIATES, INC.  
 4205 E-421-008-5  
 DATE RELEASED FOR: 4205 E-421-008-5

1 2 3 4

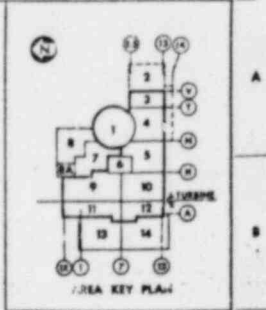
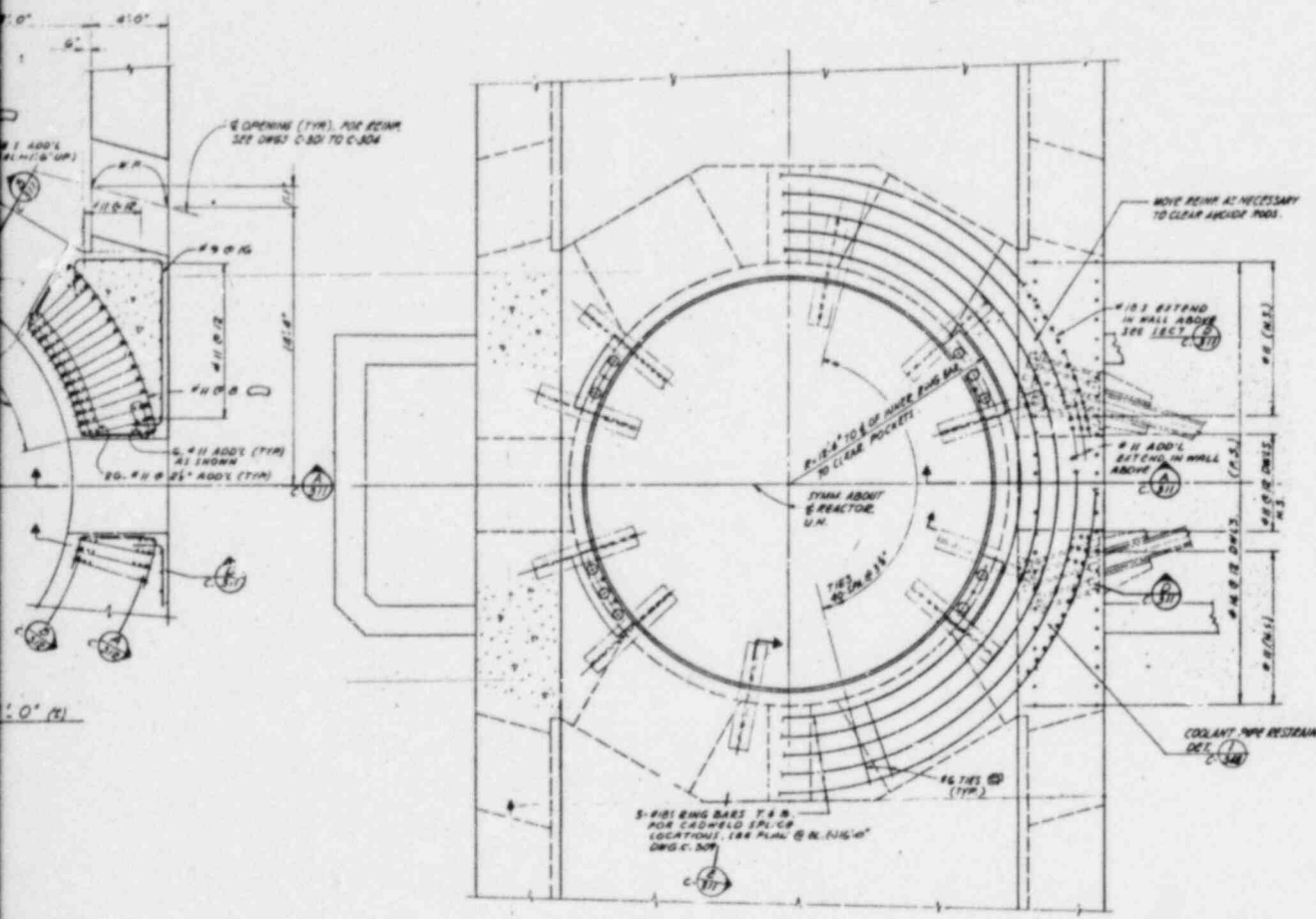


PARTIAL PLAN AT EL.

REVISION B  
ISSUED FOR INVITATION  
C/1 (2-7-68)



1 2 3 4



PLAN AT EL. 15.0' (B)

NOTES:  
 1. FOR CONC NOTES SEE DWG. C-276  
 2. WORK THIS DWG. WITH DWGS. C-309, C-311, C-312, C-400, C-401, C-281, C-348

**RECEIVED**  
 JAN 24 1979

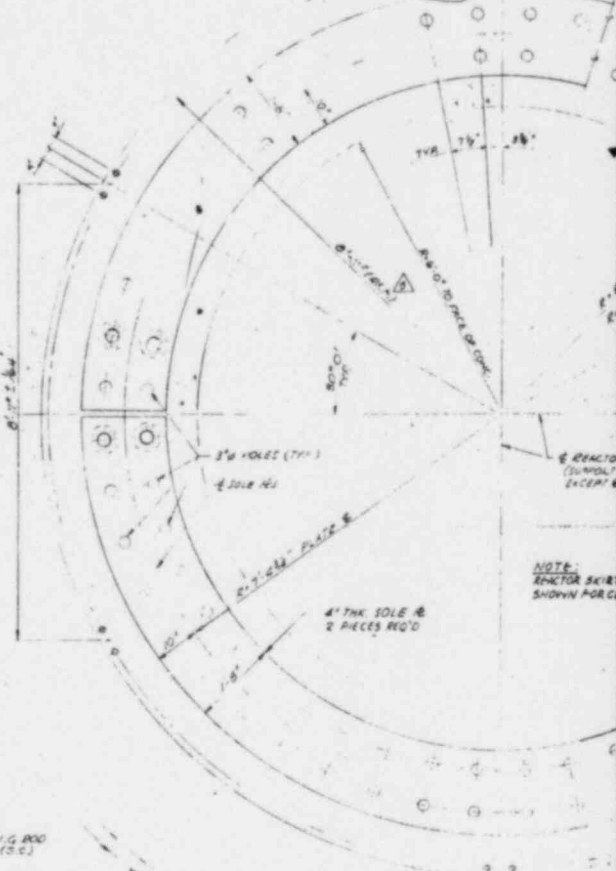
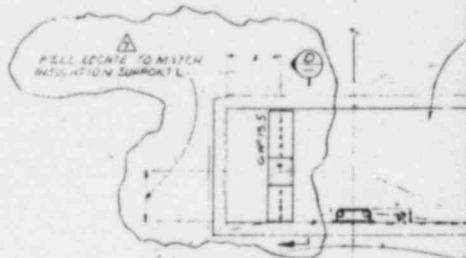
Print Control Services  
 EDS NUCLEAR INC.  
 DATUM EL. 0.0' - EL. 165.0' MSL

BECHTEL CORPORATION LOS ANGELES	
SACRAMENTO MUNICIPAL UTILITY DISTRICT SARNOFF BESS NUCLEAR STATION - UNIT 1	
REACTOR BUILDING - AREA 1 REINFORCED CONCRETE PRIMARY SHIELD - SHEET 2	
DATE: 1/23/79	SCALE: 1/8" = 1'-0"
PROJECT NO: 2392	SHEET NO: C-310

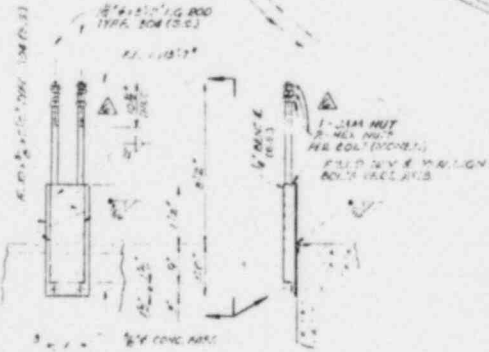
NO.	DESCRIPTION	DATE	BY	CHECKED	APPROVED
1	ADDED DWG. C-311				
2	REVISED FOR CONSTRUCTION				

ABA474

1 2 3 4 5 6



NOTE:  
REACTOR SKIRT  
SHOWN FOR CL.



PLAN  
1\"/>

SECTION (C) A

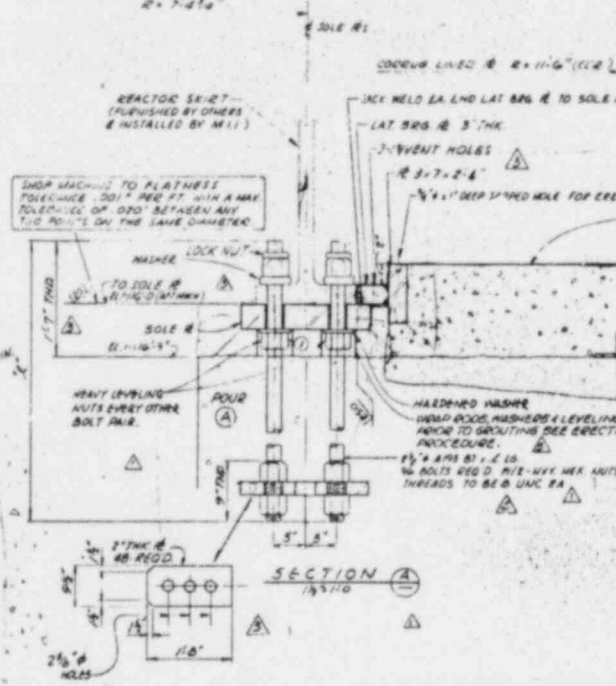
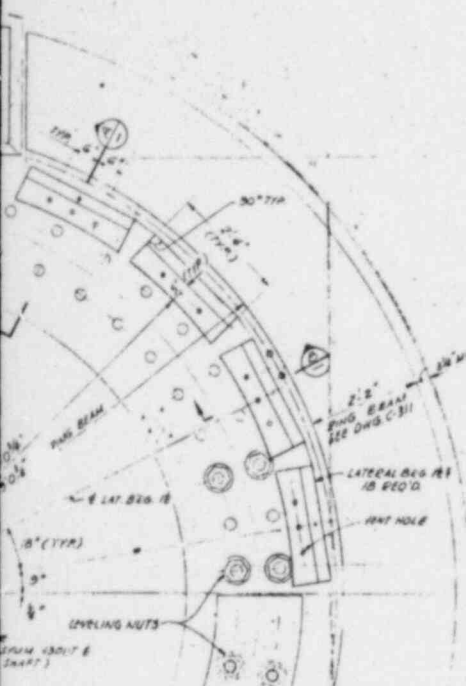
REVISION 1 ISSUED FOR  
INVITATION # 11  
ADDENDUM 2 (SEE FIELD  
ERECTOR PROCEDURE)

DATE	BY	CHKD

1 2 3 4 5 6

81

2' x 5' 0" SHAF. & EAST  
SIDE OF REACTOR ONLY



BEAM FURNISHED & CONSTRUCTED BY M.I. SEE DWG C-31 FOR DETAILS FOLLOWING PLACEMENT OF GROUT (SEE NOTE 1) AN INSPECTION SHALL BE MADE (AFTER 28 DAYS) TO ENSURE THAT NO HORIZONTAL SHRINKAGE EXISTS. IF SHRINKAGE IS EVIDENT, PRESSURE GROUT TO FILL ALL VOIDS.

2" x 8" x 8" DRILLING  
4" x 10" 5" OTHERWISE SEE DWG C-31

**WARNING:**  
NOTE 17 SHALL BE FOLLOWED FOR EACH ADJUSTMENT OF THE SUPPORT ELEV. UNDER NO CIRCUMSTANCES SHALL TURNING OF LEVELING NUTS BE ALLOWED WHILE GROUT IS ON THE SUPPORT.

**NOTES:**

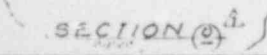
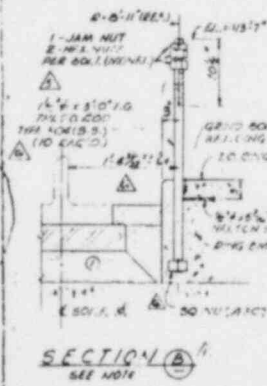
**GENERAL**

1. RANDOM SAMPLES OF EACH BOLT SIZE AND MATERIAL WILL BE SUBJECTED TO A BRINELL HARDNESS TEST, RC2 MATERIAL VERIFICATION, BY THE USER.
2. DRILL HOLES IN SOLE #2 AND MAKE A TEMPLATE OF THESE HOLES IN ACCORDANCE WITH ENGINEER'S INSTRUCTIONS TO BE USED LATER TO SPOT ANCHOR BOLTS & TO DRILL CORRESPONDING HOLES IN THE REACTOR SKIRT AS BY THE REACTOR MANUFACTURER.
3. LATRAL BEAMING #2'S SHALL BE STRESS RELIEVED AFTER WELDING & PRIOR TO MACHINING (150°F - 200°F)
4. MACHINED SURFACES SHALL BE PROTECTED FROM DAMAGE DURING SHIPMENT.
5. ALL GROUT SHALL BE NON-SHRINK (MINIMUM COMPRESSIVE STRENGTH 5000 P.S.I.) BRAND SUBJECT TO APPROVAL BY ENGINEER.
6. NORTH ARROW AND LAYER FACE SHALL BE CORRECTLY MARKED ON THE TEMPLATE.
7. ENTIRE ASSEMBLY IS (GA-1)

13. SNUG ALL LEVELING NUTS.
14. WORK TO BE DONE BY M.I.
15. ADJUST POSITION OF SOLE #2 SO THAT THE TOP FLANGE OF THE REACTOR SKIRT WILL BE LEVEL AT ELEVATION 151'-0" PER B & H AS BUILT INFORMATION.
16. ADJUST POSITION OF REACTOR IF OCCURED BY LIFTING T-2 WHEEL VERTICALLY OFF THE SUPPORT AND ADJUSTING ALL LEVELING NUTS TO INSURE FULL CONTACT WITH SOLE #2 REPEAT STEP 15.
17. REINSTALL & TIGHTEN DOWN NUTS SNUG TIGHT.
18. SNUG ALL LEVELING NUTS.
19. PLACE GROUT LEAVING SPACE AROUND LEVELING NUTS FOR VENTING.
20. AFTER GROUT HAS CURED, BACK OFF LEVELING NUTS.
21. WARP BOLTS, NUTS AND WASHERS TO PREVENT BOND AND BEARING, SEE SECT. (A).
22. PLACE REMAINING GROUT.

**FIELD ERECTION PROCEDURE:**

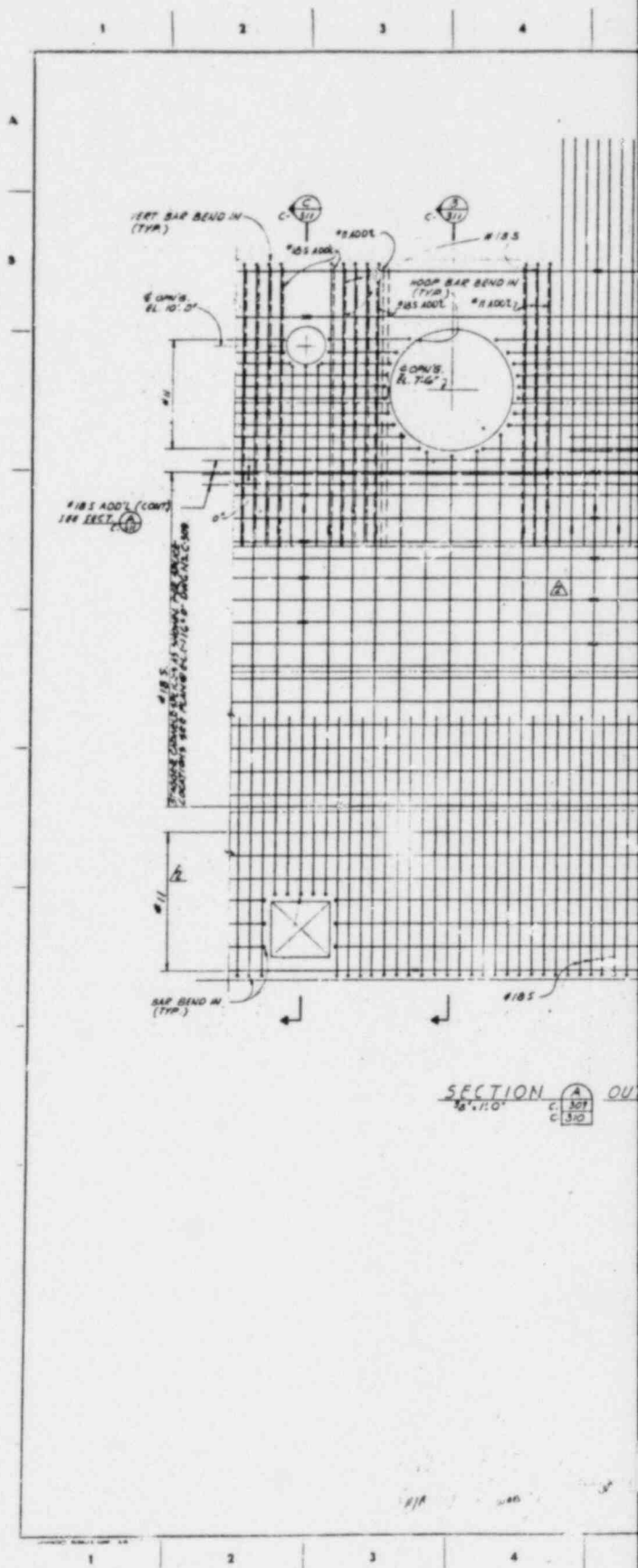
- WORK TO BE DONE BY CLE
1. PLACE CONCRETE (A) WITH ANCHOR BOLTS POSITIONED USING TEMPLATE.
  2. POSITION LEVELING NUTS SO THAT THE TOP OF THE SOLE PLATES WILL BE BELOW ELEV. 116'-0"
  3. INSTALL SOLE PLATES OVER ANCHOR BOLTS ONTO LEVELING NUTS.
  4. ADJUST LEVELING NUTS TO RAISE SOLE PLATES TO REQUIRED SETTING ELEV. WITH MAX TOLERANCE ACROSS ANY DIA OF THE SUPPORT TOP SURFACE ± 0.2"
  5. INSTALL HOLD-DOWN NUTS TIGHT.



DATUM EL. 0.0' ± EL. 165.0' MSL Print Control Services  
EDS NUCLEAR INC.

BECHTEL CORPORATION LOS ANGELES	
SACRAMENTO MUNICIPAL UTILITY DISTRICT TRENDS RECO NUCLEAR STATION - UNIT 1	
REACTOR BUILDING - AREA 1 REACTOR SUPPORT	
AB712	C-368

**RECEIVED**  
JAN 24 1979



LEFT BAR BEND IN (TYP.)

#3 ADDL

#3 ADDL

#18 S

6 CMU'S EL. 10'-0"

HOOP BAR BEND IN (TYP.)

#3 ADDL

#3 ADDL

6 CMU'S R. T.G. 2

#18 S ADDL (CONT)  
1/4" EXT.

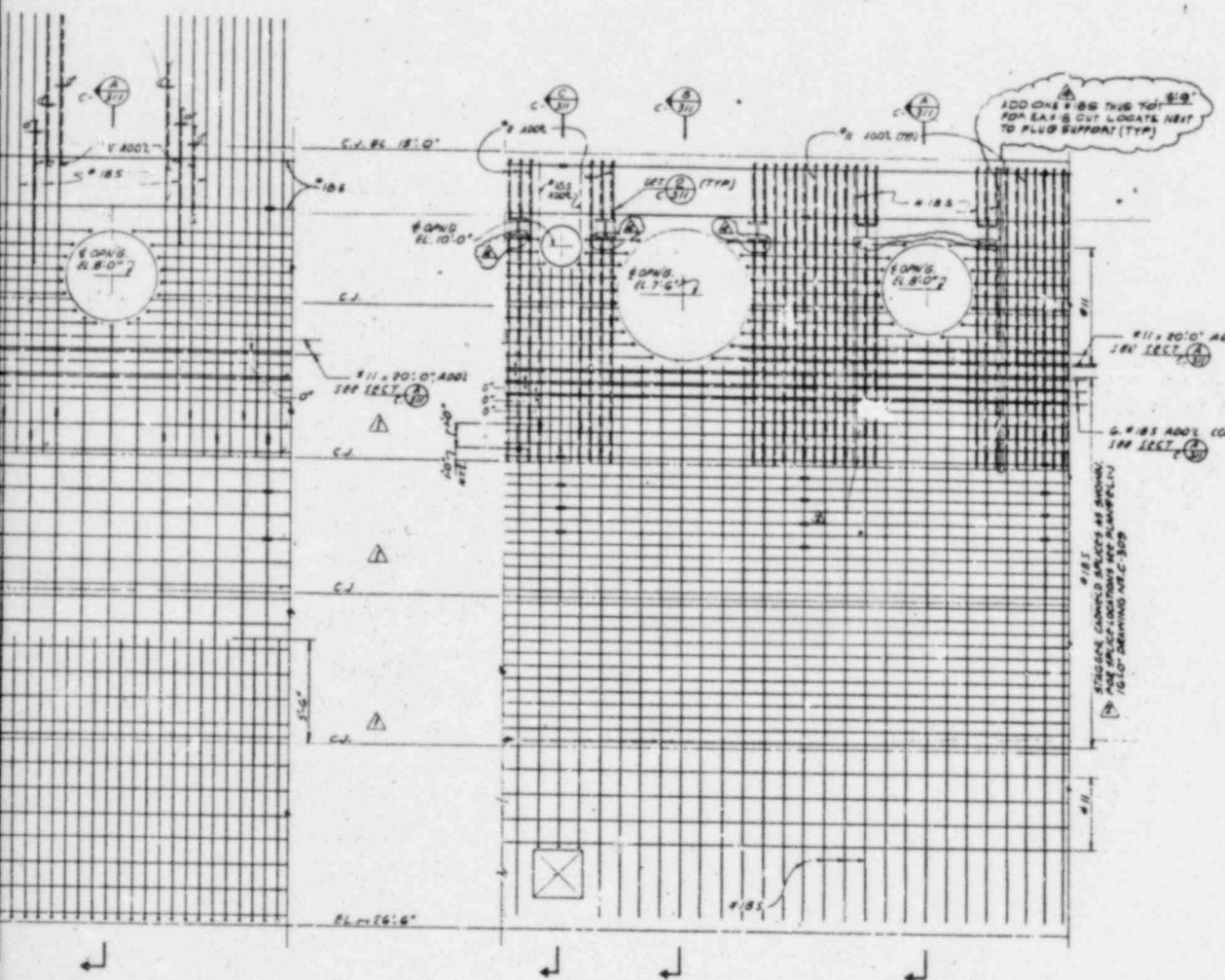
#18 S. THROUGH COLUMN AND SLAB. 1/2"

BAR BEND IN (TYP.)

#18 S

SECTION A-A  
8'-11 1/2"  
C. 307  
C. 310





INSIDE FACE

SECTION B  
 11'-0" C-312 C-310  
 INSIDE FACE

**RECEIVED**  
 JAN 24 1979

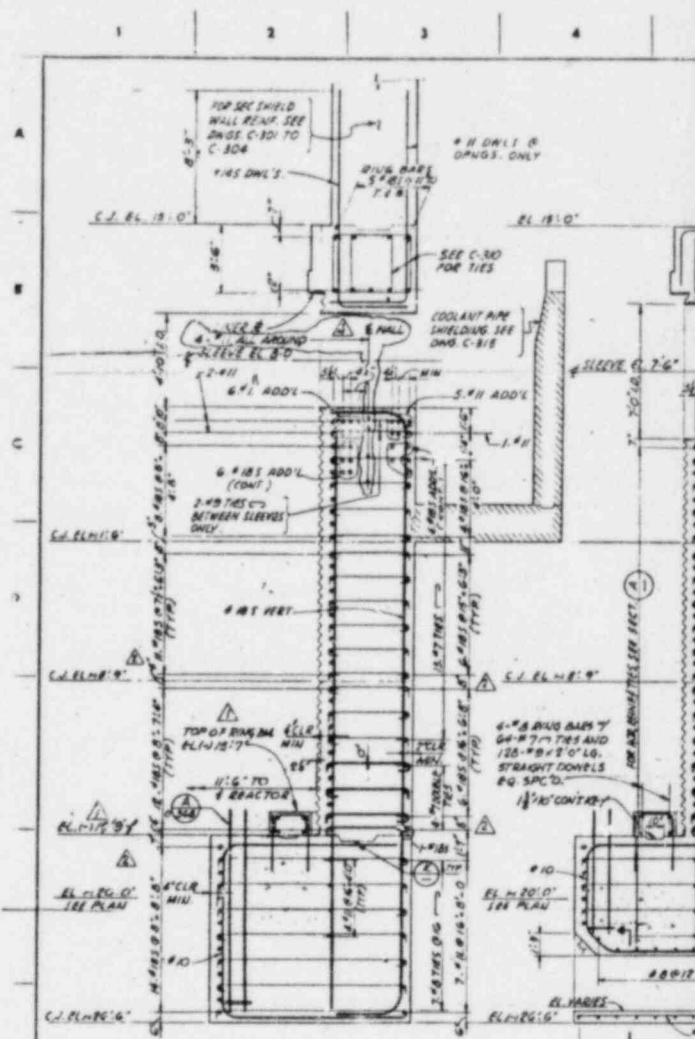
Print Control Services  
 EDS NUCLEAR INC.

NOTES:  
 1. WORK THIS DWG WITH DWGS C-309, C-310 & C-311  
 2. FOR CONCRETE NOTES SEE DWG C-270.

DATUM EL. 00' EL. 1850' WSL

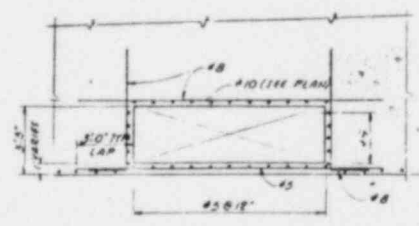
<b>BECHTEL CORPORATION</b> LOS ANGELES	
SACRAMENTO MUNICIPAL UTILITY DISTRICT <b>SANCHO SEC. NUCLEAR STATION - UNIT 1</b>	
REACTOR BUILDING - AREA I <b>REINFORCED CONCRETE</b> <b>PRIMARY SHIELD - SHEET 4</b>	
SHEET NO. <b>A183</b>	DRAWING NO. <b>C-312</b>

NO.	REVISION	DATE	BY	CHKD.	APP'D.	REMARKS
1	REMOVED CARBON UNDER PIPES					
2	REMOVED CARBON UNDER PIPES					
3	REMOVED CARBON UNDER PIPES					
4	REMOVED CARBON UNDER PIPES					
5	REMOVED CARBON UNDER PIPES					
6	REMOVED CARBON UNDER PIPES					
7	REMOVED CARBON UNDER PIPES					
8	REMOVED CARBON UNDER PIPES					
9	REMOVED CARBON UNDER PIPES					
10	REMOVED CARBON UNDER PIPES					
11	REMOVED CARBON UNDER PIPES					
12	REMOVED CARBON UNDER PIPES					
13	REMOVED CARBON UNDER PIPES					



SECTION A  
 5'-11.0"  
 C-101  
 C-102  
 C-103

SECTION B  
 5'-11.0"  
 C-104



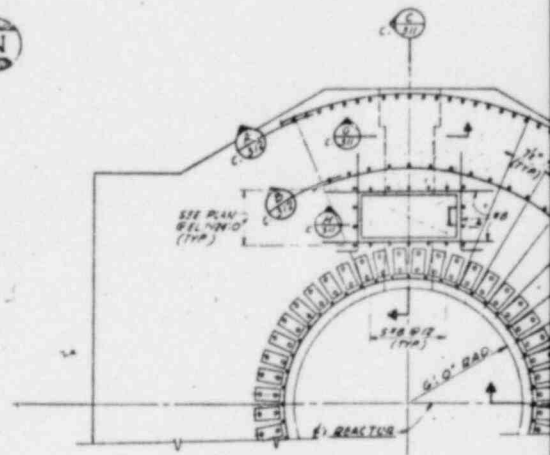
SECTION C  
 4'-11.0"  
 C-105

REVISION 2 ISSUED WITH INVITATION C-12

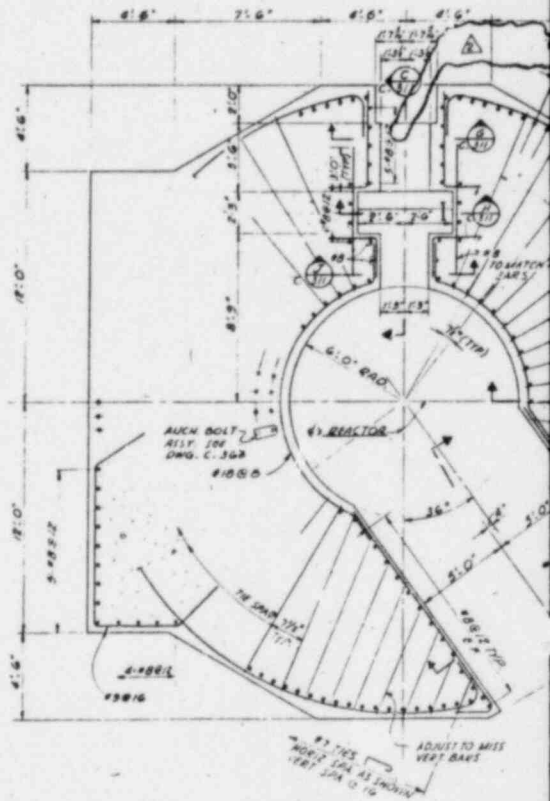
REVISION 7 ISSUED FOR INVITATION 111, ADDENDUM 2 (FOR 246 ASAM DUTY)



62



PARTIAL PLAN AT EL. 20:0

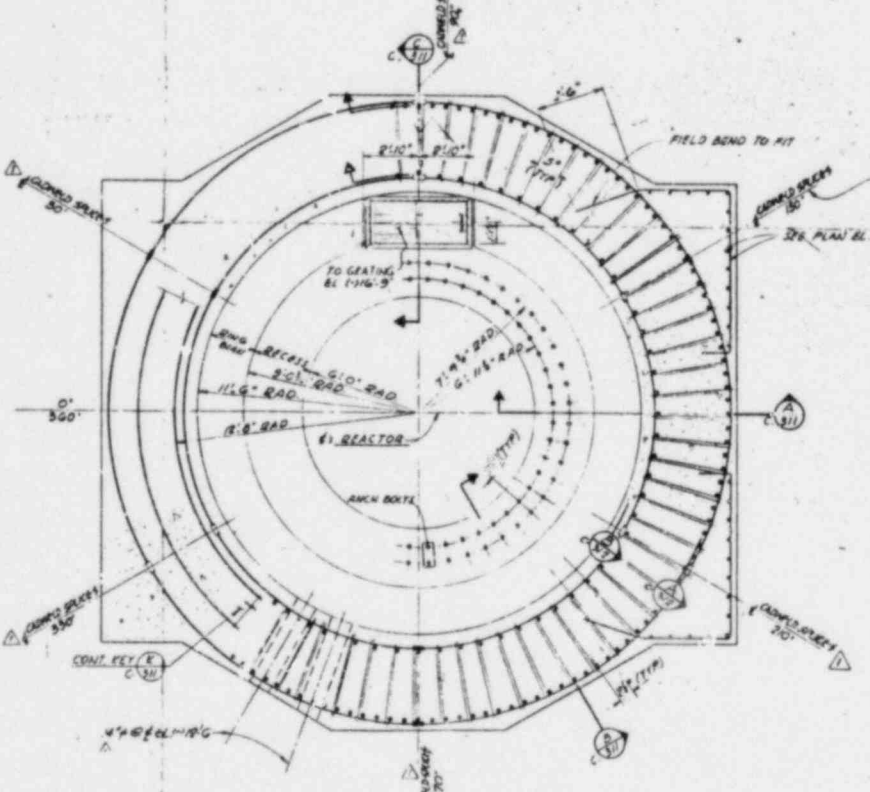
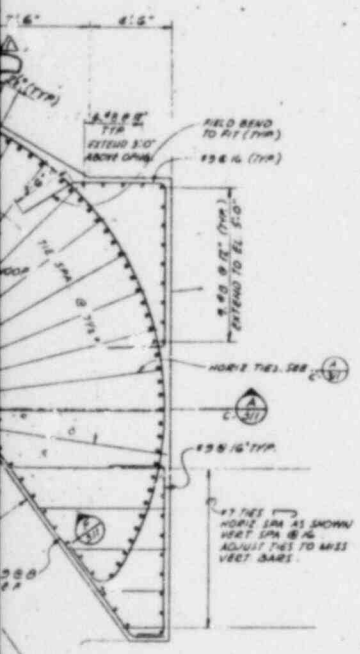
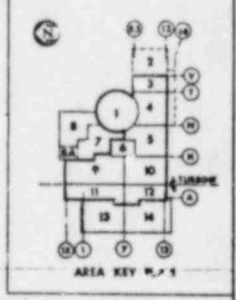
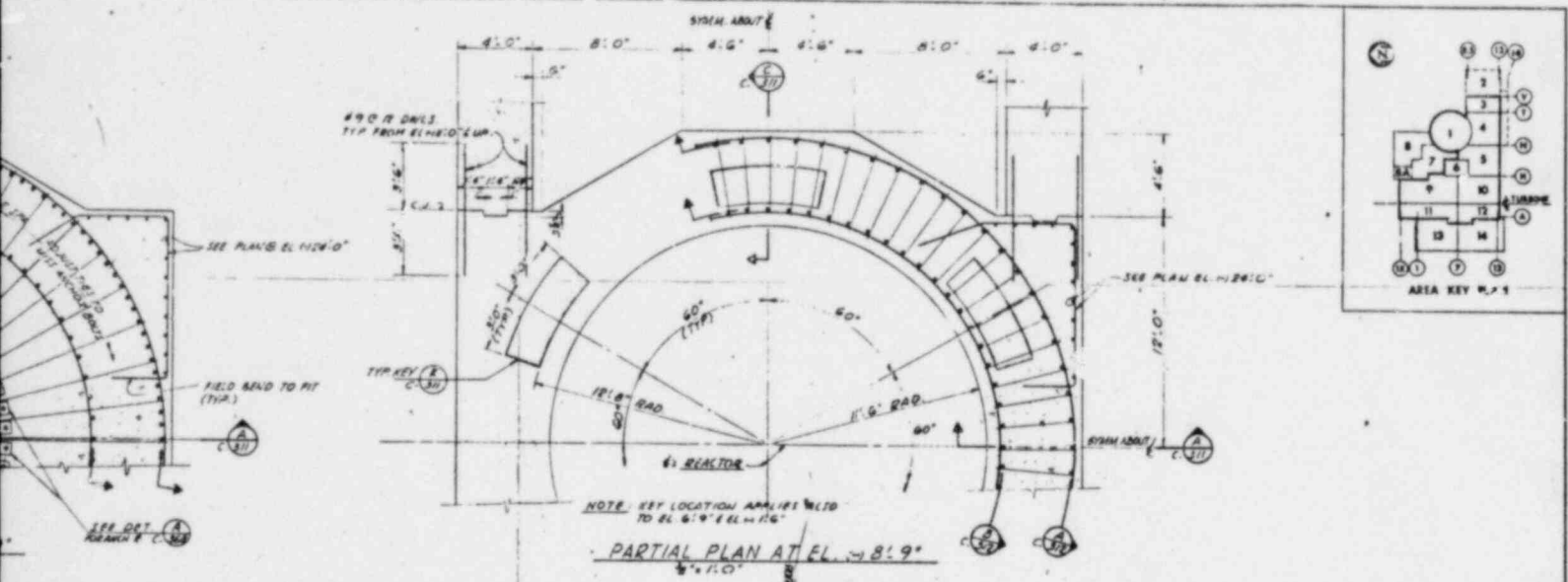


PLAN AT EL. 24:0

REVISION 1 ISSUED WITH INVITATION C.12	REVISION 2 ISSUED FOR INVITATION C.11 (2-7-69)
--	--



1 2 3 4



- NOTES**
- FOR CONC. NOTES SEE DWG. C. 276
  - WORK THIS DWG. WITH DWGS. C. 310, 311, C. 312 & 301
  - ALL REIN. FOR PRIMARY SHIELD SHALL BE WITH A G15 GRADE SO EXCEPT AS OTHERWISE NOTED.
  - ALL CONCRETE FOR PRIMARY SHIELD SHALL BE CLASS C-12 CONC. (28 DAYS STRENGTH) REFER TO DWG. C-287
  - DIMENSIONS SYMMETRICAL ABOUT  $\phi$  OF REACTOR UNLESS OTHERWISE SHOWN OR NOTED.

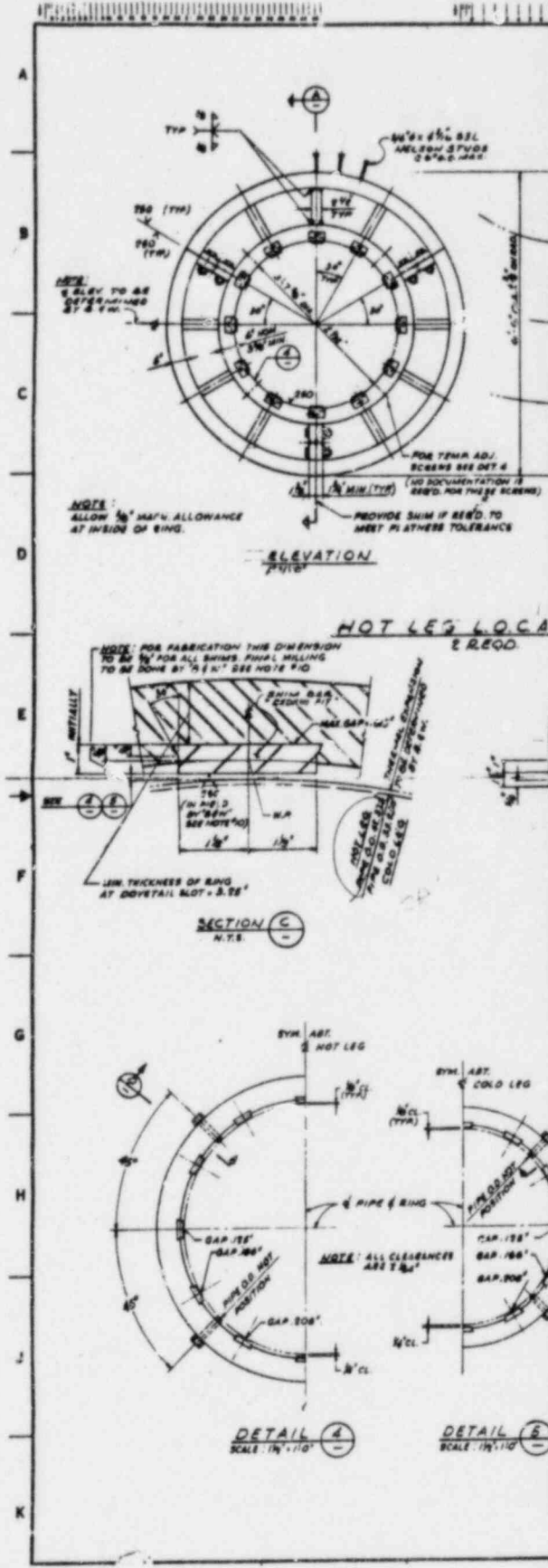
**RECEIVED**  
JAN 24 1979

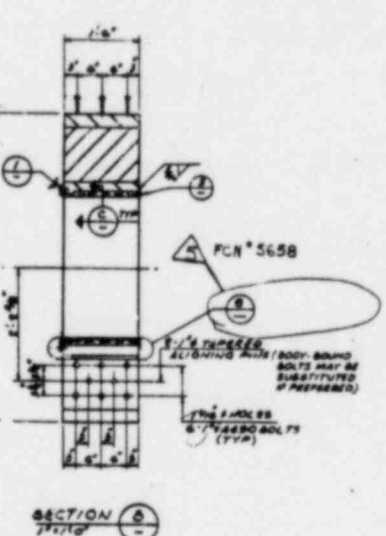
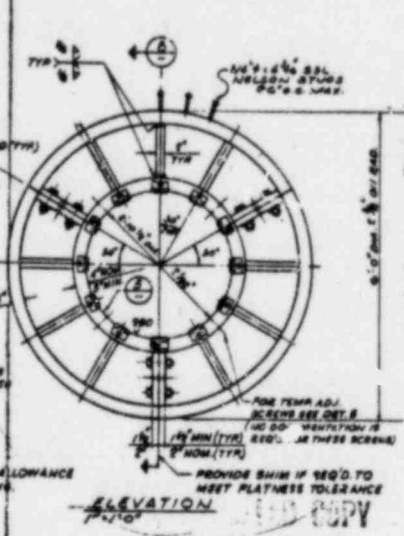
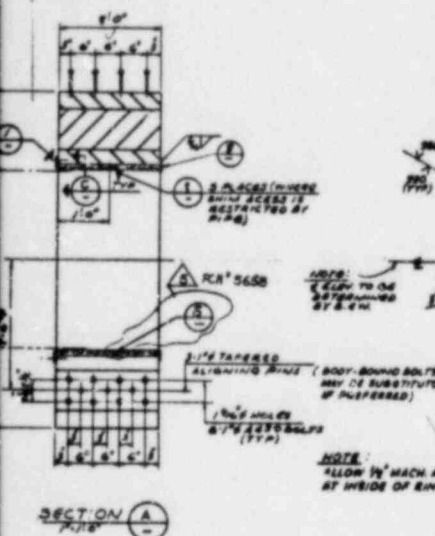
Design Control Services  
EDS NUCLEAR INC.

BECHTEL CORPORATION LOS ANGELES	
SACRAMENTO MUNICIPAL UTILITY DISTRICT TRENCH SEC'D NUCLEAR STATION - UNIT 1	
REACTOR BUILDING - AREA 1 REINFORCED CONCRETE PRIMARY SHIELD - SHEET 1	
DATE: 1/23/79	SCALE: 1/8" = 1'-0"
PROJECT NO: 8333	SHEET NO: C-309
REVISED BY: [Signature]	DATE: 1/23/79

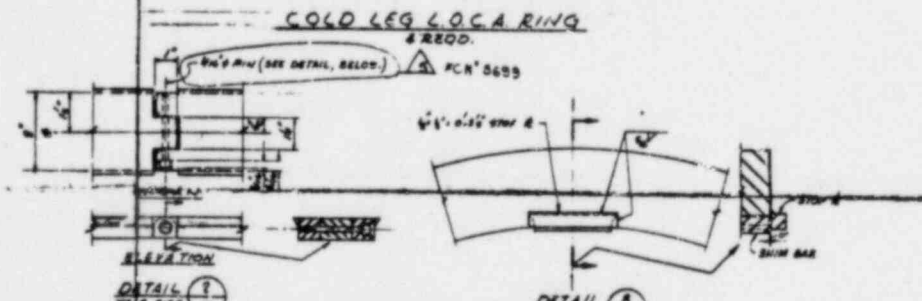
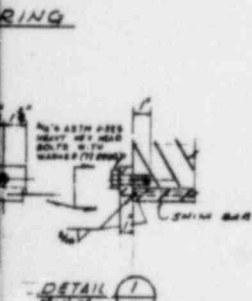
NO.	DESCRIPTION	DATE	BY	CHKD.	APP'D.
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2	CHECKER				
3	ENGINEER				
4	DESIGNER				
5	CHECKER				
6	ENGINEER				
7	DESIGNER				
8	CHECKER				
9	ENGINEER				
10	DESIGNER				
11	CHECKER				
12	ENGINEER				
13	DESIGNER				

ABA586

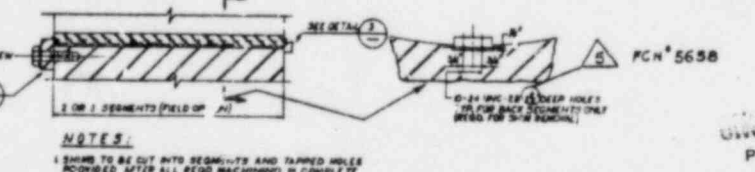
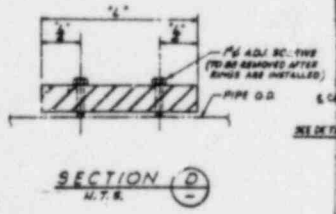
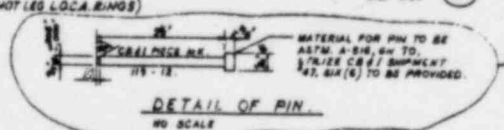




- NOTES:**
1. ALL STRUCTURAL STEEL AND WELDING METALS SHALL BE IN ACCORDANCE WITH SPEC. NO. 1743-C-198A.
  2. ALL STRUCTURAL STEEL PLATE GROUP ON THE DRAWING SHALL CONFORM TO JET/...-...
  3. ALL BOLTS SHOWN ON THE DRAWING SHALL CONFORM TO ASTM...
  4. ALL BOLTS SHOWN ON THE DRAWING SHALL CONFORM TO...
  5. ALL FINISH SURFACES EXCEPT WHERE NOTED SHALL BE SURFACES SO THERE IS NO MORE THAN .001" TOLERANCE ACROSS FINISH SURFACES.
  6. UNFINISHED SURFACES SHALL BE USED UNDER ALL BUTS AND BOLT HEADS.
  7. ...
  8. EACH L.O.C.A. RING STRAIN SHALL BE PROVIDED WITH STREETS, RELIEVED AS PER SPEC. NO. 1743-C-198A PARAGRAPH 5.8.
  9. EACH L.O.C.A. RING IS MADE OUT OF (2) THICK WELDED STEEL SHEETS & MANUFACTURED AND STREETS, RELIEVED THE RING SHALL BE ASSEMBLED AND THEN MACHINED INSIDE SURFACE GROUNDS FOR APPROX. THE SHIM BAR AS SPECIFIED.
  10. ALL HOLES INVOLVING BRACKETING AND POSITIONING THE L.O.C.A. RINGS TOGETHER WITH ALL HELD MILLING OF DOW-TAIL SHIM BARS TO PROVIDE REQUIRED CLEARANCES AROUND THE PIPE SHALL NOT TO BE DONE BY "BRACING" OR "WELDED".
  11. MATCH MARK ALL PARTS IN A GIVEN ASSEMBLY.
  12. MACHINING ALLOWANCE: ON BOLTED INTERFACES SHALL BE .01" ON T.O. OF COLD LEG L.O.C.A. RING SHALL BE .01"
- NOTE A:**  
 DWS. C-198A REV. A HAS FORMED 19-06-73  
 DWS. C-198A REV. A HAS FORMED 19-06-73



- REFERENCE DWS**
- C-198 CONTAINMENT INTERNAL STRUCTURES REACTIVE SHELD WALL SHEET 1
  - C-201 CONTAINMENT INTERIOR STRUCTURES REACTIVE SHELD WALL SHEET 2
  - C-198 CONTAINMENT INTERNAL STRUCTURES REACTIVE SHELD WALL SHEET 3



- NOTES:**
1. SHIMS TO BE CUT INTO SEGMENTS AND TAPPED HOLES PROVIDED AFTER ALL REED MACHINING IS COMPLETE.
  2. THE 1/2" DIA. UNF 28 (1) DEEP HOLES MAY ACCED SO THAT A THREADED ROD MAY BE USED TO REMOVE SHIMS.

**UNCONTROLLED COPY  
 POWER ENGINEERING**

**RECEIVED**  
 JAN 25 1979

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 EDS NUCLEAR INC.

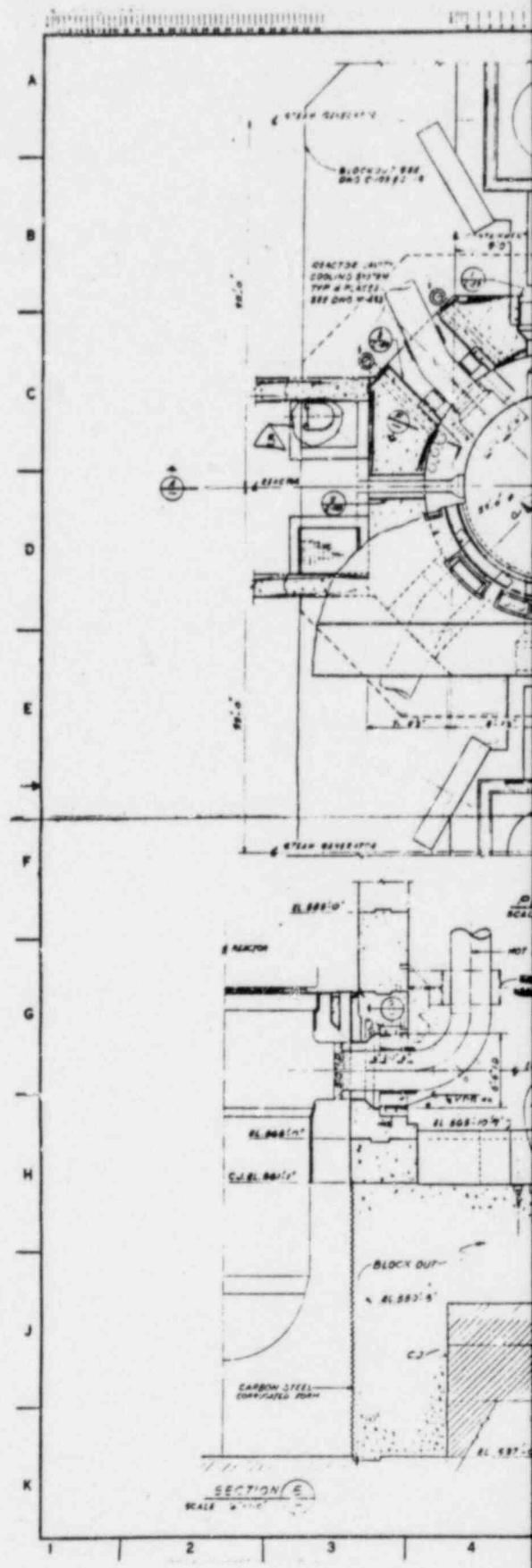
NO.	DESCRIPTION	DATE	BY	CHKD.
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3	ADDED FCN 5658	1/25/79	J	J
4	ADDED FCN 5658	1/25/79	J	J
5	ADDED FCN 5658	1/25/79	J	J
6	ADDED FCN 5658	1/25/79	J	J
7	ADDED FCN 5658	1/25/79	J	J
8	ADDED FCN 5658	1/25/79	J	J
9	ADDED FCN 5658	1/25/79	J	J
10	ADDED FCN 5658	1/25/79	J	J
11	ADDED FCN 5658	1/25/79	J	J
12	ADDED FCN 5658	1/25/79	J	J
13	ADDED FCN 5658	1/25/79	J	J
14	ADDED FCN 5658	1/25/79	J	J

**REGTEL COMPANY**  
 1401 W. 10th Street  
 Tulsa, Oklahoma 74106

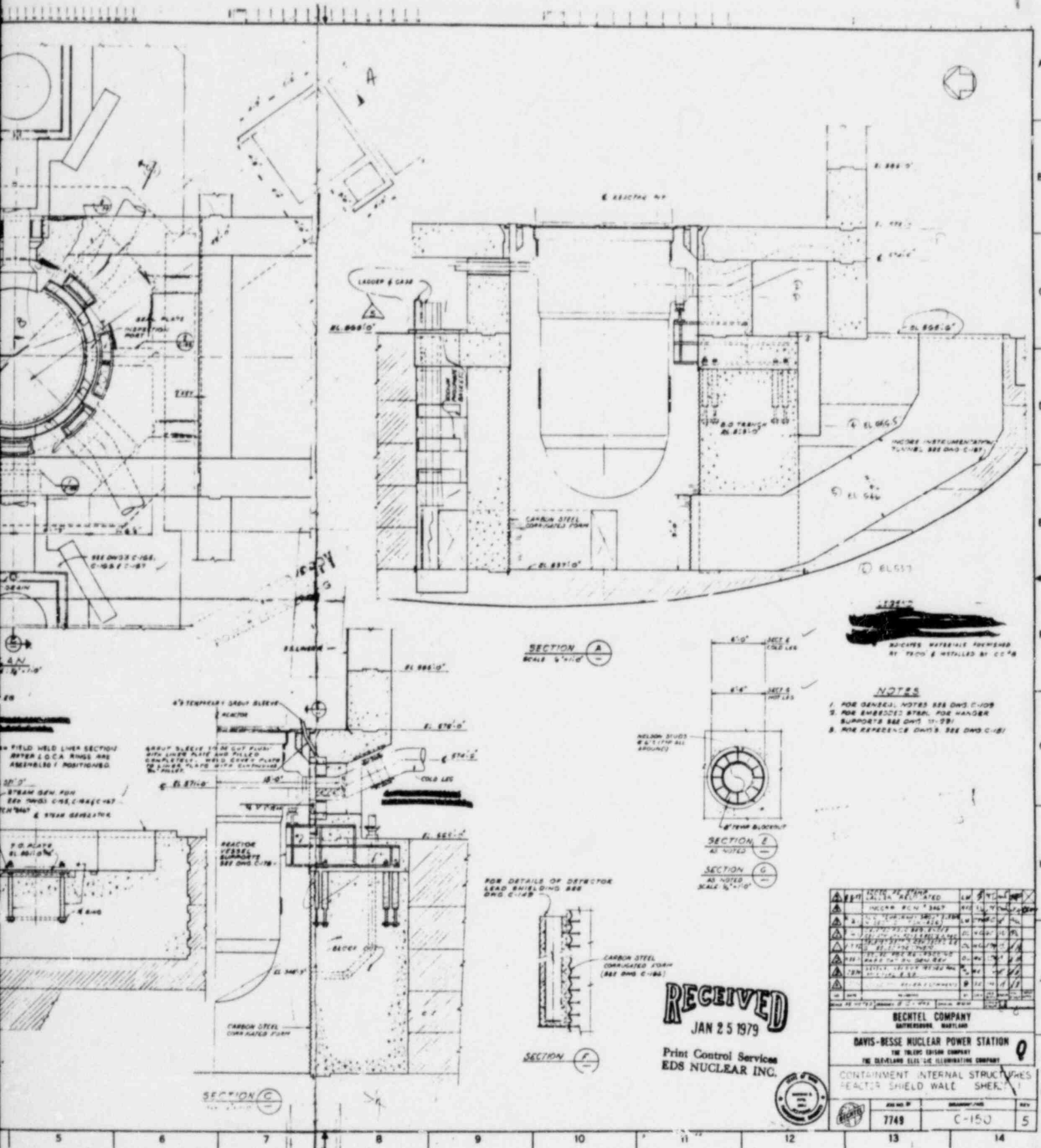
**DAYIS-BESSE NUCLEAR POWER STATION**  
 THE OKLAHOMA ELECTRIC CORPORATION

**CONTAINMENT INTERNAL STRUCTURES  
 REACTOR COOLANT LOOP  
 L.O.C.A. RESTRAINTS**

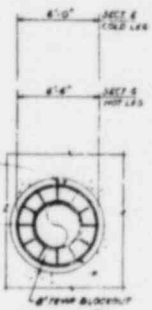
REV. NO.	7748	ISSUED BY	C-198A	REV.	5
----------	------	-----------	--------	------	---







SECTION A  
SCALE 5/8"=1'-0"



SECTION E  
AS NOTED

SECTION G  
AS NOTED  
SCALE 5/8"=1'-0"

FOR DETAILS OF DETECTOR  
LEAD SHIELDING SEE  
DWG C-148



SECTION F  
SCALE 5/8"=1'-0"

~~REDACTED~~  
INDICATE MATERIALS FURNISHED  
BY TRADE & INSTALLED BY CONTRACTOR

**NOTES**

1. FOR DETAILS, NOTES SEE DWG C-108
2. FOR EMBEDDED STEEL, FOR HANGER  
SLANGERS SEE DWG 11-127
3. FOR REFERENCE DIMS. SEE DWG C-187

NO.	DATE	BY	CHKD.	APP.	DESCRIPTION
1	11/15/77	J. W. ...	J. W. ...	J. W. ...	ISSUED FOR CONSTRUCTION
2	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
3	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
4	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
5	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
6	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
7	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
8	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
9	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
10	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
11	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
12	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
13	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION
14	11/15/77	J. W. ...	J. W. ...	J. W. ...	REVISION

**RECEIVED**  
JAN 25 1979

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EDS NUCLEAR INC.

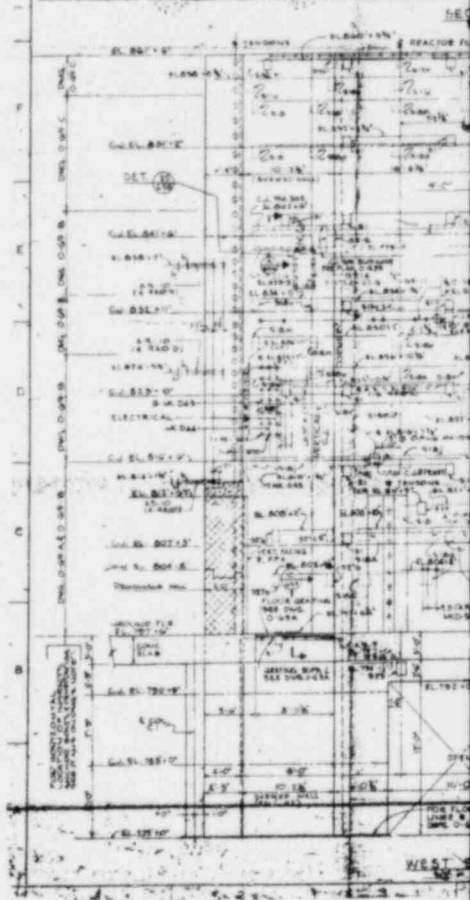
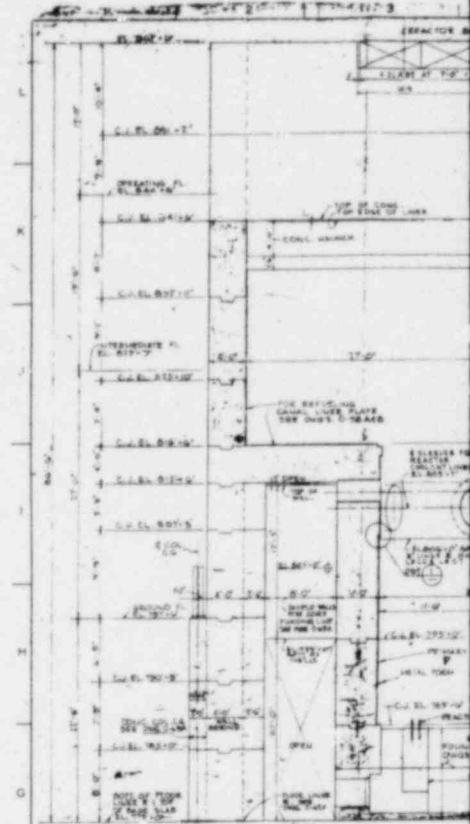


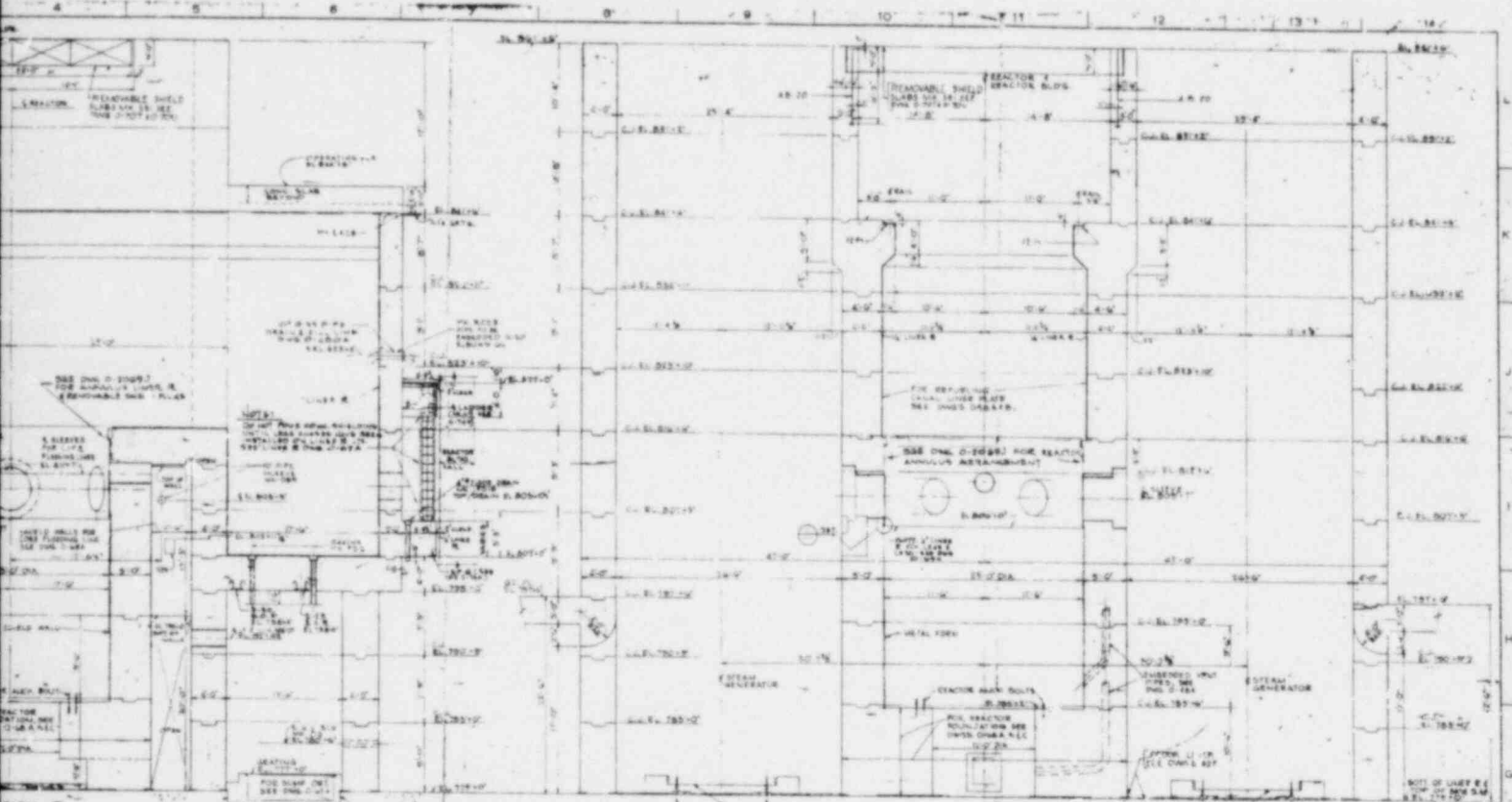
**BECTEL COMPANY**  
DALLAS, TEXAS

**DAVIS-BESSE NUCLEAR POWER STATION**  
THE BREVILLE (DAVIS) COMPLEX  
THE DALLAS-BESSE LLC ILLUMINATING COMPANY

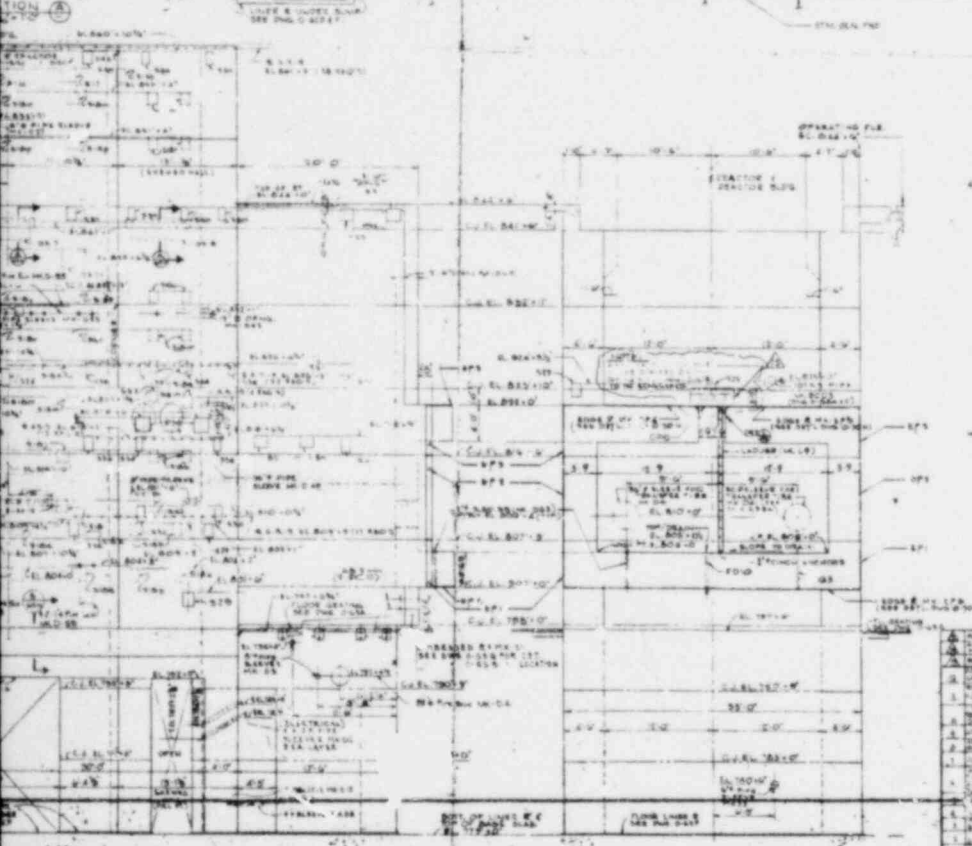
CONTAINMENT - INTERNAL STRUCTURES  
REACTOR SHIELD WALL SHEET

7748 C-150 5





SECTION 1-10



SECTION 1-11

SECTION 1-12

SECTION 1-13

SECTION 1-14

SECTION 1-15

SECTION 1-16

SECTION 1-17

SECTION 1-18

SECTION 1-19

SECTION 1-20

SECTION 1-21

SECTION 1-22

SECTION 1-23

SECTION 1-24

SECTION 1-25

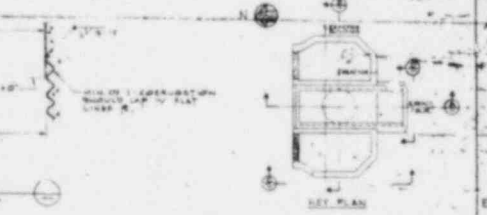
SECTION 1-26

SECTION 1-27

SECTION 1-28

SECTION 1-29

SECTION 1-30



DETAIL 1-1

DETAIL 1-2

DETAIL 1-3

DETAIL 1-4

DETAIL 1-5

DETAIL 1-6

DETAIL 1-7

DETAIL 1-8

DETAIL 1-9

DETAIL 1-10

DETAIL 1-11

DETAIL 1-12

DETAIL 1-13

DETAIL 1-14

DETAIL 1-15

NOTE:  
FOR OTHER NOTES AND REVISIONS, SEE DWG. 4-2A.  
SEE PLANS FOR DIMENSIONS, LOCATIONS OF WALLS, REINFORCEMENTS AND ANCHOR BOLTS.  
REVISIONS TO THE DRAWING OF ARCHIVE  
DATE OF REVISION: 10/18/55  
BY: [Signature]

SECTION II  
(REACTOR ANNULUS ENLARGEMENT)

UNIT 1 ONLY  
JAN 18 1955  
Reactor File Control  
KELLOGG, INC.

DATE: [Signature]  
DATE ENGINEER APPROVED: [Signature]

REACTOR CORP. JOB NO. 620

DAY POWER COMPANY

OCONEE NUCLEAR STATION UNITS 1 & 2

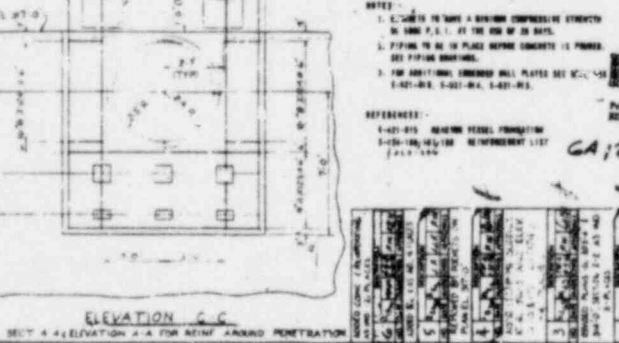
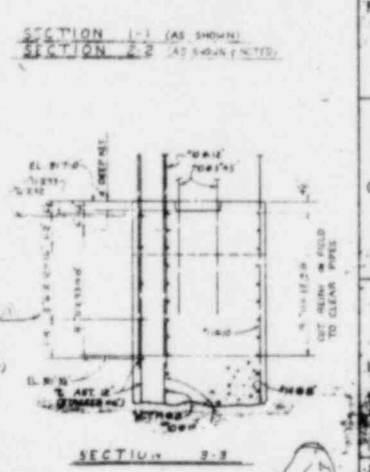
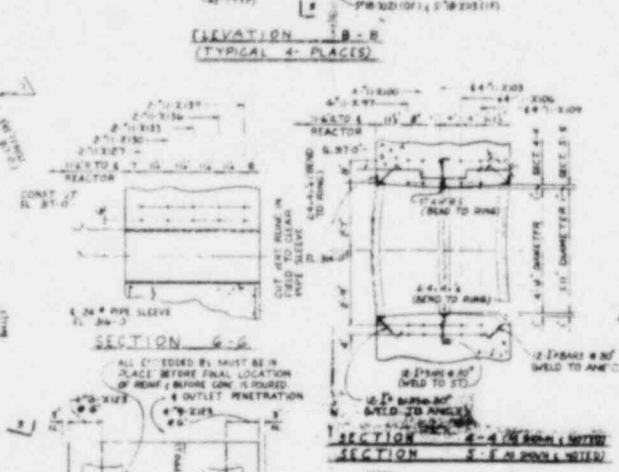
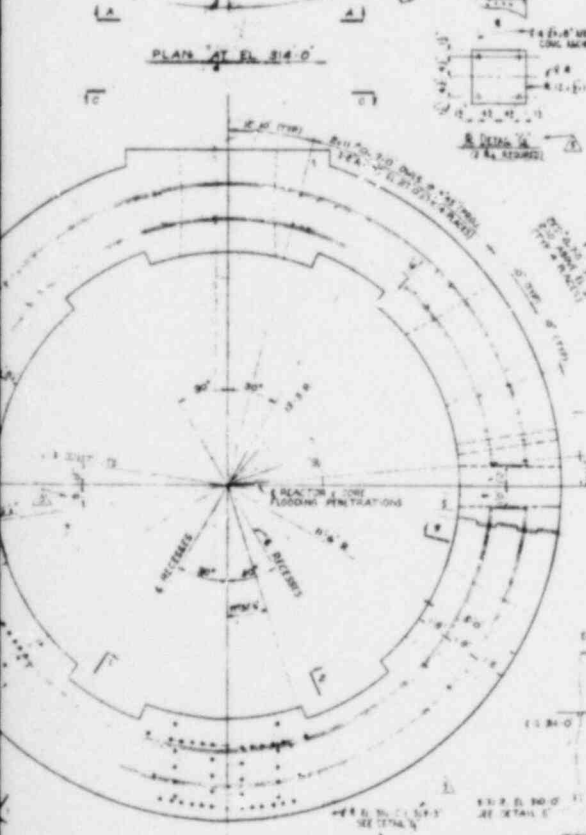
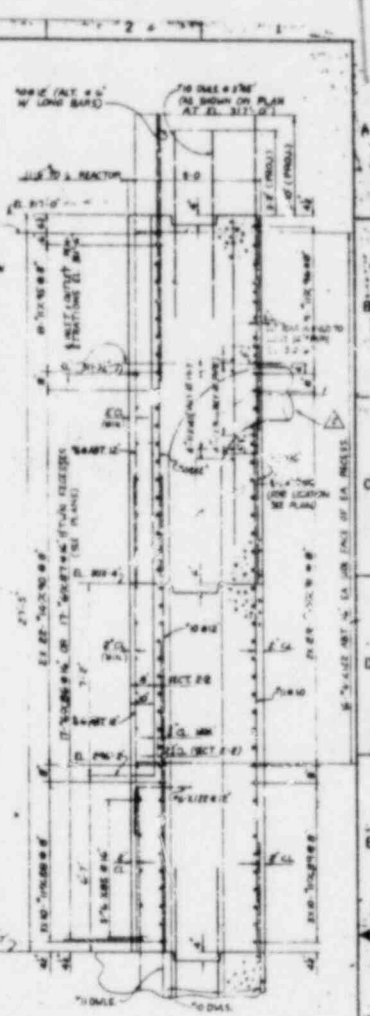
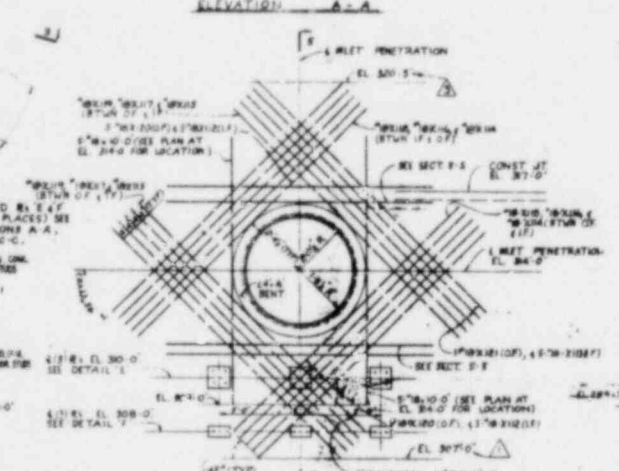
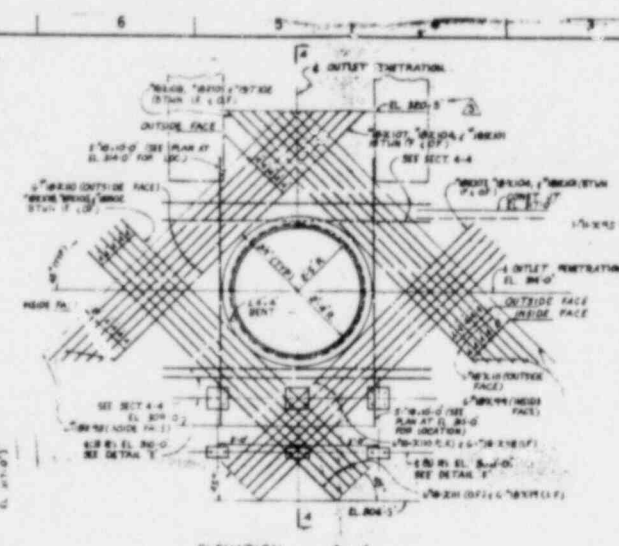
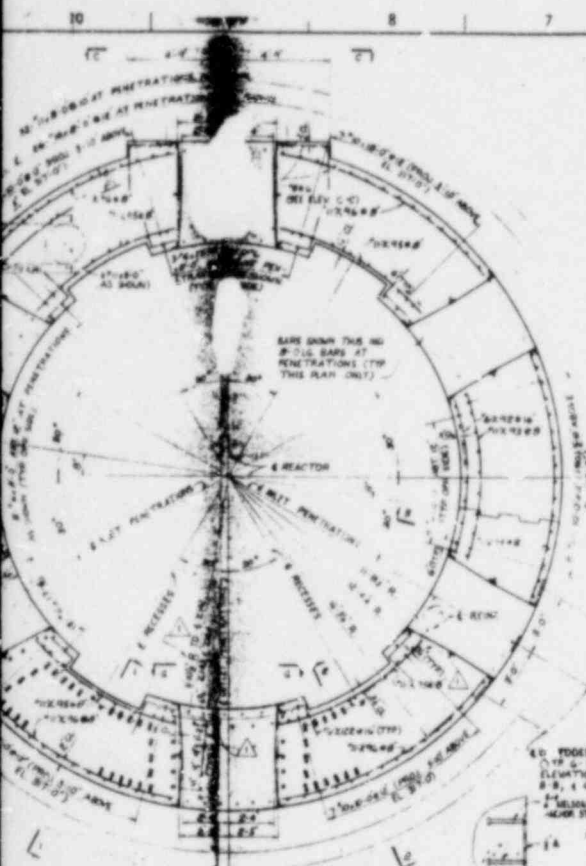
REACTOR BUILDING  
PRIMARY & SECONDARY SHIELD WALLS

SECTIONS & ELEVATIONS  
CONCRETE

NO.	DESCRIPTION	DATE	BY	CHKD.
1	ISSUED FOR CONSTRUCTION	10/18/55	[Signature]	[Signature]
2	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
3	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
4	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
5	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
6	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
7	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
8	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
9	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]
10	REVISIONS TO THE DRAWING OF ARCHIVE	10/18/55	[Signature]	[Signature]

SOUTH ELEV. 1-1





- NOTES:
1. CONCRETE TO HAVE A MINIMUM COMPRESSIVE STRENGTH OF 4000 P.S.I. AT THE END OF 28 DAYS.
  2. REINFORCEMENT TO BE IN PLACE BEFORE CONCRETE IS Poured. SEE FILING DRAWINGS.
  3. FOR ADDITIONAL UNBARRED WALL PLATES SEE E.C. 11-1-61-01-1, 1-01-01-1, 1-01-01-2.

REFERENCES:  
 1-01-01-01 REACTOR VESSEL TRANSMISSION  
 1-01-01-02 REACTOR VESSEL TRANSMISSION LIST  
 1-01-01-03

RECEIVED	DATE	RELEASED FOR	CHAP
NOV 18 1961			
METROPOLITAN Edison COMPANY			
THREE MILE ISLAND NUCLEAR STATION UNIT 1			
REACTOR BUILDING-CONCRETE			
REACTOR VESSEL PRIMATE HALL			
ELEV. 288'-7" TO ELEV. 317'-0"			
GLIBERT ASSOCIATES, INC.			
DESIGNERS AND CONSULTANTS			
NO. 4192	E-421-016		

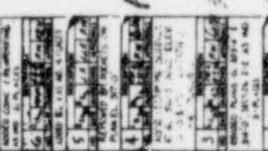
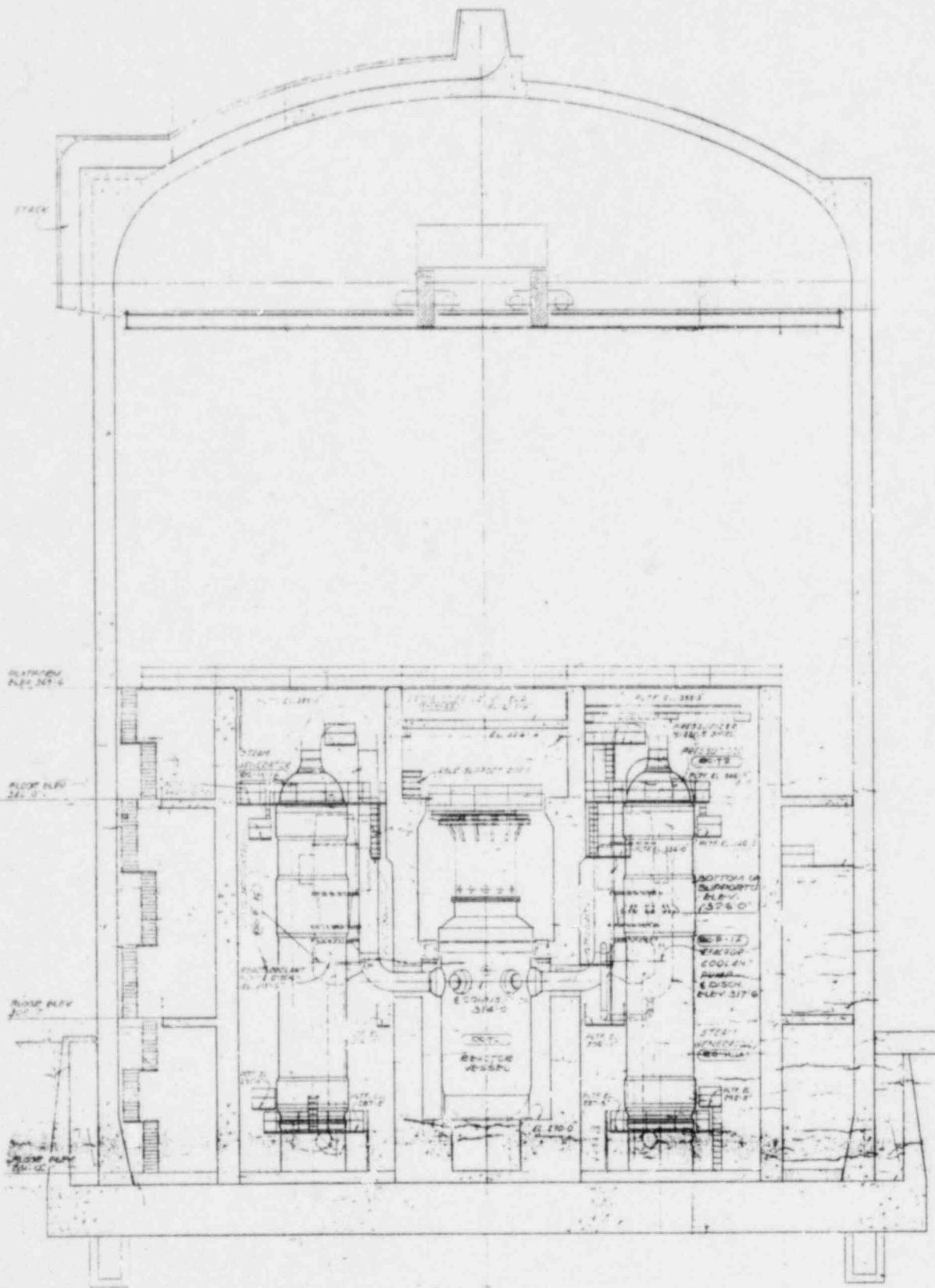


Figure TMI-1. Containment (Dwg E-002-004)



APPENDIX B  
Typical Pressure-Time Histories  
For Cavity Wall Analysis





\*\*\*\*\*  
+ PROGRAM MAP +  
+ VERSION 09/20/79 +  
\*\*\*\*\*

22 1.000  
 23 1.000  
 24 1.000  
 25 1.000  
 26 1.000

# SECTION 1: INPUT DATA

IM# 0 (NO, NO WRITE ONTO TAPE 2, #1, WRITE ALL T/H'S, #2, WRITE ONLY MERGED TIME HISTORIES  
 IP# 0 (NO, NO PUNCHED DECKS, #1, PUNCH ALL T/H'S, #2, PUNCH ONLY MERGED TIME HISTORIES)

TOTAL AREAS OF EDS PRESSURE TIME HISTORIES, INCHES SQUARE

7874.1600	7874.1600	7874.1600	7874.1600	7874.1600	7874.1600	5129.0400	5129.0400
5129.0400	5129.0400	5129.0400	5129.0400	5129.0400	5129.0400	4840.0800	4840.0800
4840.0800	5020.6800	1267.1600	1337.7600	1337.7600	1337.7600	1287.1600	1287.1600
1061.7600	941.7600	941.7600	941.7600	941.7600	941.7600	1287.1600	1287.1600
1337.7600	1337.7600	1337.7600	1247.1600	1247.1600	3589.3600	3408.7600	3408.7600
3408.7600	3589.3600	1247.1600	1247.1600	1247.1600	3589.3600	3408.7600	3408.7600

Surface area of EDS-SNAP BRICK ELEMENTS

NUMBER OF SUB AREAS FOR EACH EDS PRESSURE TIME HISTORY

2	2	2	2	2	2	2	2
3	2	2	2	2	2	2	2
4	3	2	2	2	2	2	2
2	4	4	4	4	4	2	2
3	4	2	2	2	2	2	2

NUMBER OF CRAFT VOLUMES OVERLAPPING ON EDSSNAP ELEMENTS

APPROXIMATE DIRECTION ANGLE FOR EACH EDS PRESSURE TIME HISTORY

165.0000	135.0000	105.0000	75.0000	75.0000	75.0000	165.0000	165.0000
105.0000	75.0000	45.0000	15.0000	15.0000	15.0000	105.0000	105.0000
45.0000	15.0000	165.0000	135.0000	135.0000	135.0000	45.0000	45.0000
150.6850	141.3550	98.6450	81.3550	81.3550	21.3550	165.0000	165.0000
105.0000	75.0000	45.0000	15.0000	15.0000	15.0000	105.0000	105.0000

SUBAREA VALUES FOR EACH EDS PRESSURE TIME HISTORY WITH CORRESPONDING BW VOLUME

EDS P, T/H, 1 AREA #	7874.1600	SUM OF ALL SUB AREAS #	7874.1600
BW VOL#	10	1119.7200	22
EDS P, T/H, 2 AREA #	7874.1600	SUM OF ALL SUB AREAS #	7874.1600
BW VOL#	11	1119.7200	23
EDS P, T/H, 3 AREA #	7874.1600	SUM OF ALL SUB AREAS #	7874.1600
BW VOL#	12	1119.7200	24
EDS P, T/H, 4 AREA #	7874.1600	SUM OF ALL SUB AREAS #	7874.1600
BW VOL#	1	1119.7200	13
EDS P, T/H, 5 AREA #	7874.1600	SUM OF ALL SUB AREAS #	7874.1600
BW VOL#	2	1119.7200	14

BREAK DOWN OF WHICH BW CRAFT VOLUMES ARE ON EACH EDS EDSSNAP ELEMENT WITH SQUARE INCHES.

EDS P, T/H, 6 AREA # 7874.1600, SUM OF ALL SUB AREAS # 7874.1600  
 BW VOL# 3, 15

6750.0000	1119.7200				
8M VOL#	3	15			
EDS P. T/H.	7 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	22	34			
EDS P. T/H.	8 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	23	35			
EDS P. T/H.	9 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	24	36			
EDS P. T/H.	10 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	13	25			
EDS P. T/H.	11 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	14	26			
EDS P. T/H.	12 AREA #	5129.00000	SUM OF ALL SUB AREAS#		5129.00000
4081.5000	1047.0000				
8M VOL#	15	27			
EDS P. T/H.	13 AREA #	5020.00000	SUM OF ALL SUB AREAS#		5020.00000
4153.0000	478.5000	300.2000			
8M VOL#	34	46	47		
EDS P. T/H.	14 AREA #	4840.00000	SUM OF ALL SUB AREAS#		4840.00000
4153.0000	343.1000	343.1000			
8M VOL#	35	47	40		
EDS P. T/H.	15 AREA #	4840.00000	SUM OF ALL SUB AREAS#		4840.00000
4153.0000	343.1000	343.1000			
8M VOL#	36	48	37		
EDS P. T/H.	16 AREA #	4840.00000	SUM OF ALL SUB AREAS#		4840.00000
4153.0000	343.1000	343.1000			
8M VOL#	25	37	36		
EDS P. T/H.	17 AREA #	4840.00000	SUM OF ALL SUB AREAS#		4840.00000
4153.0000	343.1000	343.1000			
8M VOL#	26	38	39		
EDS P. T/H.	18 AREA #	5020.00000	SUM OF ALL SUB AREAS#		5020.00000
4153.0000	300.2000	478.5000			
8M VOL#	27	39	40		
EDS P. T/H.	19 AREA #	1287.10000	SUM OF ALL SUB AREAS#		1287.10000
465.4300	821.7300				
8M VOL#	46	47			

8M VOL# 46

EDS P. T/H. 20 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
866.8800 470.8800  
8M VOL# 47

EDS P. T/H. 21 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
866.8800 470.8800  
8M VOL# 37

EDS P. T/H. 22 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
866.8800 470.8800  
8M VOL# 37

EDS P. T/H. 23 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
470.8800 866.8800  
8M VOL# 34

EDS P. T/H. 24 AREA # 1287.16000 . SUM OF ALL SUB AREAS# 1287.16000  
821.7300 465.4300  
8M VOL# 39

EDS P. T/H. 25 AREA # 1061.76000 . SUM OF ALL SUB AREAS# 1061.76000  
97.4400 433.4400 97.4400 433.4400  
8M VOL# 58

EDS P. T/H. 26 AREA # 941.76000 . SUM OF ALL SUB AREAS# 941.76000  
433.4400 37.4400 433.4400 37.4400  
8M VOL# 59

EDS P. T/H. 27 AREA # 941.76000 . SUM OF ALL SUB AREAS# 941.76000  
37.4400 433.4400 37.4400 433.4400  
8M VOL# 60

EDS P. T/H. 28 AREA # 941.76000 . SUM OF ALL SUB AREAS# 941.76000  
433.4400 37.4400 433.4400 37.4400  
8M VOL# 49

EDS P. T/H. 29 AREA # 941.76000 . SUM OF ALL SUB AREAS# 941.76000  
37.4400 433.4400 37.4400 433.4400  
8M VOL# 50

EDS P. T/H. 30 AREA # 941.76000 . SUM OF ALL SUB AREAS# 941.76000  
57.4400 433.4400 37.4400 433.4400  
8M VOL# 51

EDS P. T/H. 31 AREA # 1287.16000 . SUM OF ALL SUB AREAS# 1287.16000  
465.4300 821.7300  
8M VOL# 58

EDS P. T/H. 32 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
866.8800 470.8800  
8M VOL# 59

EDS P. T/H. 33 AREA # 1337.76000 . SUM OF ALL SUB AREAS# 1337.76000  
866.8800 470.8800  
8M VOL# 60

EDS P. T/H.	33 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	49	470.8800		
		60		
EDS P. T/H.	34 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	49	470.8800		
		50		
EDS P. T/H.	35 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	51	470.8800		
		50		
EDS P. T/H.	36 AREA	1287.16000	SUM OF ALL SUB AREAS	1287.16000
8M VOL	51	465.6300		
		52		
EDS P. T/H.	37 AREA	3589.38000	SUM OF ALL SUB AREAS	3589.38000
8M VOL	80	1270.9500		
		81	568.8900	478.5900
			58	59
EDS P. T/H.	38 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	81	1270.9500		
		82	433.4400	433.4400
			59	60
EDS P. T/H.	39 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	82	1270.9500		
		71	433.4400	433.4400
			60	49
EDS P. T/H.	40 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	71	1270.9500		
		72	433.4400	433.4400
			49	50
EDS P. T/H.	41 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	72	1270.9500		
		73	433.4400	433.4400
			50	51
EDS P. T/H.	42 AREA	3589.38000	SUM OF ALL SUB AREAS	3589.38000
8M VOL	73	1270.9500		
		74	478.5900	568.8900
			51	52

EDS PRESSURE FUNCTIONS 43, 44, 45, 46 NOT SHOWN IN THIS RUN, BUT ARE  
 EQUAL TO CORRESPONDING KRAFT VOLUMES DATA IN PENETRATIONS  
 (TWO HOTTER AND TWO COLDER PENETRATIONS)

# SECTION 2: CRAFTZ VOLUME DATA

CASE# 1 20A HC

TIME (SECONDS)

PRESSURE (PSIG) OF CRAFTZ VOLUME

LOAD STEP #	0.0050	0.0100	0.0150	0.0200	0.0250	0.0300	0.0350	0.0400	0.0450	0.0500
LOAD STEP # 1	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000	0.0000
LOAD STEP # 41	.0003	.0002	.0007	.0006	.0002	-.0001	-.0001	.0002	.0002	.0002
LOAD STEP # 101	.0021	.0118	.0634	.0629	.0115	.0014	.0006	.0009	.0012	.0012
LOAD STEP # 107	.0602	.2902	.7503	.7443	.2874	.0584	.0090	.0031	.0029	.0031
LOAD STEP # 112	.7657	2.0451	3.3644	3.3399	2.0300	.7392	.1665	.0299	.0090	.0092
LOAD STEP # 117	3.8499	6.5794	8.3451	8.3202	6.5146	3.5495	1.3830	.3579	.0775	.0740

B-8

EDS P. T/H.	33 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	470.8800	60		
	49			
EDS P. T/H.	34 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	470.8800	50		
	49			
EDS P. T/H.	35 AREA	1337.76000	SUM OF ALL SUB AREAS	1337.76000
8M VOL	470.8800	50		
	51			
EDS P. T/H.	36 AREA	1287.16000	SUM OF ALL SUB AREAS	1287.16000
8M VOL	465.4300	52		
	51			
EDS P. T/H.	37 AREA	3589.38000	SUM OF ALL SUB AREAS	3589.38000
8M VOL	1270.9500	58		
	60	81		
EDS P. T/H.	38 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	1270.9500	62		
	81			
EDS P. T/H.	39 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	1270.9500	60		
	82			
EDS P. T/H.	40 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	1270.9500	60		
	82			
EDS P. T/H.	41 AREA	3408.78000	SUM OF ALL SUB AREAS	3408.78000
8M VOL	1270.9500	50		
	72			
EDS P. T/H.	42 AREA	3589.38000	SUM OF ALL SUB AREAS	3589.38000
8M VOL	1270.9500	74		
	73			

EDS PRESSURE FUNCTIONS 43, 44, 45, 46 NOT SHOWN IN THIS RUN, BUT ARE  
 EQUAL TO CORRESPONDING CRAFTZ VOLUMES DATA IN PENETRATIONS  
 (TWO HOTLES AND TWO COLD LEE PENETRATIONS)





## LOAD STEP # 117

.0250

3.8000	6.5794	8.3451	8.3202	6.5186	3.6495	1.3430	.3579	.0775	.0780
3.572	1.0589	8.3716	12.0286	15.4055	15.4178	11.9338	8.3036	4.4133	1.5332
3.397	3.398	1.5312	4.4429	10.6806	18.6739	48.0447	48.2717	16.4355	10.5736
6.1318	2.7694	.M02M	.M0H2	2.7527	6.1362	7.7663	25.6792	56.6014	525.5371
95.9610	23.4792	7.7345	4.1373	1.4161	.6266	1.4110	4.1422	8.0519	27.3403
103.8574	466.4991	103.0990	27.1341	8.0191	3.3375	1.1010	.3673	1.0987	3.4044
277.8953	.1280	9.4511	9.3384	1.6621	1.6664	3.8438	3.8519	.0952	.0071
8.0391	23.8208	98.4748	420.6115	97.7877	23.6377	8.0017	3.1080	.0323	.2609
.A324	3.1183	.3788							

## LOAD STEP # 122

.0300

9.7896	12.9486	15.1183	15.0895	12.7225	9.0457	5.3061	2.2414	.7002	.7107
2.3397	5.7554	11.8561	17.0913	23.2800	23.2210	16.8999	11.6709	6.2764	4.9448
2.0722	2.0761	4.9787	8.3799	13.1678	33.7580	73.6091	73.7550	33.2394	13.0667
9.0750	5.7960	3.1947	3.2092	5.7924	9.1296	11.9690	43.9424	153.5343	706.9244
152.5938	43.5614	11.8889	6.6765	3.9075	2.6778	3.8994	6.6915	12.1201	44.7130
154.5384	623.2044	153.5484	44.4065	12.4075	6.0766	3.3625	1.7309	3.3569	6.0895
418.7875	.7029	19.5137	19.3227	3.0801	3.0833	5.2859	5.2594	.2219	.0137
12.6207	45.0852	155.0230	587.1469	154.0405	44.7539	12.5585	6.1208	2.8097	1.4588
2.4065	6.1325	2.0155							

## LOAD STEP # 127

.0350

16.4704	20.9035	24.1802	23.9894	20.2052	15.3501	11.0759	7.2605	3.8456	3.9409
7.7620	12.3447	16.9397	26.6693	32.5434	32.4675	26.1512	16.4080	11.1355	8.5226
6.4150	6.4527	8.7364	11.5184	19.7464	56.6185	99.4275	100.3125	55.1327	19.3779
12.2981	8.9134	7.3082	7.3356	8.9768	12.4373	17.8609	63.3905	211.7656	860.0615
210.5939	62.8510	17.712	10.3367	7.2579	6.3697	7.2688	10.3958	19.3252	68.4818
210.9968	753.4709	209.8231	68.0463	19.0371	9.4252	6.7250	4.7516	6.7261	9.2646
578.7401	2.3690	35.3397	35.0266	4.5047	4.5192	8.2839	8.2483	.4347	.0238
18.2976	70.3675	214.0485	756.1804	212.8508	69.9212	18.1702	9.3681	6.3042	4.4916
6.2996	9.3902	5.8500							

## LOAD STEP # 132

.0400

24.5244	30.6771	33.6646	33.3294	29.5911	23.3296	17.1594	14.0882	12.0703	12.4449
15.2147	19.0262	27.9273	38.5985	38.5486	38.6289	37.3787	26.8170	17.4931	12.8130
12.0876	12.2456	13.4208	18.5594	31.5209	86.1456	108.3790	109.1477	83.8721	30.4559
19.3180	13.6689	12.1211	12.1885	13.9175	19.9230	26.5262	82.8371	259.8834	900.7467
258.6162	81.9233	26.1279	19.4237	11.5523	10.7029	11.5983	15.6710	30.2008	94.9361
266.5253	807.7471	265.2176	94.3686	29.9767	11.2267	10.6061	9.0271	10.6072	11.3015
735.9943	5.1216	58.2495	57.7167	6.7630	6.1476	14.1163	14.2961	.7456	.0385
29.4150	94.1151	262.0462	886.5742	260.7966	93.5879	29.0845	11.5389	10.6592	9.1133
10.6502	11.4492	11.1331							

## LOAD STEP # 138

.0456

36.3119	41.9702	42.9129	42.8480	40.8726	34.3248	27.4386	23.9684	25.3547	26.0006
25.4462	29.1062	41.4942	46.9252	41.4280	40.3232	45.7570	39.8857	31.1158	24.5160
20.8467	21.3000	25.8254	32.7683	43.7063	109.8735	106.3790	104.7942	108.4193	42.6821
32.3414	23.4203	19.2605	19.5105	24.1722	33.7075	47.0164	109.6551	300.0149	889.9567
298.8501	108.1553	46.1364	17.0293	17.1193	16.9663	17.1406	17.5360	42.3707	119.8892
308.8119	863.3084	307.5568	118.7487	41.5021	17.1194	16.1498	14.4676	16.1808	16.9651
852.5682	10.9581	42.9619	42.2535	11.3160	10.2940	23.7368	23.4187	1.2140	.0603
44.2206	120.0918	303.8203	955.8957	302.4904	119.4393	43.7736	15.6305	15.2786	14.4009
15.3086	15.7106	19.7605							

## LOAD STEP # 143

.0506

47.5452	48.8888	48.7578	48.2949	47.7302	45.0553	41.2725	38.7361	36.6042	36.8688
39.7651	43.6143	47.5473	46.9484	45.6685	45.0798	41.0318	46.1948	43.7215	39.1133
35.0908	35.5304	40.5836	45.3889	50.4146	118.1194	107.8649	106.5048	116.8964	49.2447
44.4672	35.2215	30.2038	30.8342	36.6055	45.8625	37.6371	131.0987	326.6595	870.4009
325.7070	129.7139	57.0688	29.0661	24.4263	25.6808	24.9099	30.1330	59.2209	136.1924
333.8203	925.6789	332.6707	137.2504	58.1864	24.1635	22.0951	20.6720	22.2124	24.4902
903.4113	18.8079	101.8695	100.8698	16.8836	16.7743	36.2748	35.5336	1.7465	.0829
55.0055	137.4871	338.2477	961.6273	336.8672	136.5817	54.2490	24.1399	21.7336	19.9039
21.0992	24.1362	30.6329							

## LOAD STEP # 148

.0556

55.1467	52.8549	53.1406	53.0227	51.1398	52.1761	55.6730	55.8256	51.2871	51.3486
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## LOAD STEP # 148

.055b

55,1467	52,8549	53,1406	53,0227	51,1398	52,1761	55,6730	55,8256	51,2871	51,3486
56,9555	58,2651	51,1740	49,2071	52,4593	53,2552	48,0880	49,6061	52,1355	52,8335
52,4516	52,4193	53,6451	53,5338	54,9960	120,1872	114,5212	114,5831	119,5781	53,5993
50,3811	49,2146	47,4412	47,7898	54,3074	51,0022	62,9542	144,1733	307,1575	65,6259
346,3037	143,0575	62,3943	44,2937	38,0738	39,7076	38,7141	51,0330	69,6850	154,2745
356,2672	963,7189	355,1426	153,2428	68,7824	55,1752	32,1628	20,7372	32,9866	35,6117
920,001	27,2196	118,4098	117,3463	24,8758	25,6427	50,3297	44,5496	2,3829	,1075
71,138	149,3700	361,0265	948,0172	350,8158	146,3162	69,5668	33,9636	30,3986	28,1291
30,7311	34,5386	46,0191							

## LOAD STEP # 153

.067b

61,0078	57,2409	57,2407	57,0129	55,3290	58,9192	64,9655	68,2542	71,6002	72,0476
69,6835	66,8096	58,2877	56,2034	59,7738	59,9042	55,3158	56,4518	59,0846	64,5977
67,1029	67,1077	65,1170	60,5459	50,2326	123,4035	123,2876	123,7347	122,8276	56,9160
57,2428	62,7452	66,7875	64,6371	63,2254	58,6418	67,7654	150,7830	365,6912	851,6274
364,3517	169,8913	66,5117	59,3722	55,3138	58,3257	35,5337	59,7578	70,7945	163,0315
372,1647	970,5131	371,1928	162,0104	70,3435	51,1030	49,8351	43,2663	50,5612	50,7965
913,9634	38,7055	129,4998	129,0711	36,8450	38,1066	57,2869	56,7118	3,1129	,1340
79,5932	157,9662	373,3507	941,8916	372,3620	156,9677	79,1321	48,4254	45,7515	40,6022
46,9230	50,1901	62,7275							

## LOAD STEP # 158

.065b

67,9111	64,4080	62,2905	61,2617	63,2772	68,3630	71,3965	78,5757	88,4057	88,6937
78,9493	70,9323	67,6452	66,1430	65,9809	64,8004	64,7453	66,3748	70,2390	75,5810
62,2392	62,5930	76,5188	71,3767	64,1848	128,1928	131,4776	131,3726	127,9664	62,7405
68,2897	75,3858	80,9990	81,0270	75,5916	69,2840	72,2688	155,3864	382,4261	860,4914
381,2063	154,8971	71,3937	65,3619	69,0317	73,0471	68,9228	65,9889	73,0022	165,2831
379,7768	956,8840	379,0184	164,4572	71,9299	66,6185	68,9184	62,5804	69,5723	67,6967
893,8623	54,6784	135,8814	135,6494	49,9362	50,9665	57,9214	57,4330	3,9258	,1623
70,8426	160,4597	381,3308	948,6859	380,5334	150,6567	70,8417	68,3910	69,8536	62,8565
70,8703	70,2544	77,3017							

## LOAD STEP # 163

.070b

75,4374	73,1115	69,4402	68,8982	74,1562	78,5828	82,4959	89,8304	94,7588	94,2911
88,1856	79,1370	77,4451	76,1576	71,3574	70,4821	73,5403	77,2459	82,7222	88,8621
95,981	96,0488	89,2553	82,5926	76,0556	132,5429	134,1556	137,7941	133,5840	74,4715
80,2033	85,7714	90,5200	90,9048	86,6354	80,5893	77,1209	158,8677	393,1892	877,9146
392,4241	158,3914	76,6079	81,4344	82,1513	82,9857	82,8108	82,2001	74,1215	163,7616
385,6005	943,3737	384,9785	163,0782	73,5311	79,0596	86,1222	86,2382	86,3622	79,8000
868,4908	71,1141	137,7406	137,1091	60,2255	60,9258	58,4505	47,7832	4,8146	,1922
67,4031	155,9881	386,5642	957,2751	385,9259	155,4479	66,6567	80,3105	89,2795	87,4571
89,3356	80,4834	87,5625							

## LOAD STEP # 168

.075b

85,6329	83,3810	79,6478	80,8803	85,2356	90,2754	94,1855	99,2991	96,6341	95,5983
96,5397	93,1499	91,5485	95,0368	79,3055	78,8802	91,2922	91,1770	95,5271	101,8326
101,1439	100,6760	100,1040	93,5847	104,7933	104,6185	142,2729	142,7591	111,8357	104,4432
94,9520	98,2177	100,3695	100,4629	96,9203	95,5934	86,2240	156,7896	398,4017	896,7089
398,2266	156,9273	85,5343	92,7551	97,8898	97,1712	98,4021	93,1441	77,7165	159,3204
394,3197	944,0835	393,7348	158,9771	77,0600	86,3435	94,4516	102,6733	99,6618	87,9905
847,6367	86,4409	137,2262	136,8387	70,6402	71,0312	59,2460	58,6881	5,7823	,2240
76,3752	149,3037	390,6112	957,9470	390,0998	149,0062	75,4728	85,0922	97,2645	103,6504
96,9288	84,8280	94,7587							

## LOAD STEP # 173

.080b

102,6231	96,4471	94,4375	96,0375	99,9007	106,0611	109,5838	106,9529	102,0900	100,8406
103,4375	106,0812	111,8751	98,5901	104,6115	105,0413	104,2624	114,0951	111,7330	108,2891
103,0057	102,0921	105,7083	109,6149	113,7176	89,1831	124,0803	126,5965	89,2698	117,7443
115,9727	113,0766	111,1774	110,7421	111,9462	115,0145	107,9902	130,9071	402,5356	910,9657
401,9982	133,7980	106,6766	103,6785	108,4251	110,1156	107,8573	103,8971	94,6973	150,2537
402,7445	953,7620	402,2074	151,1617	93,1119	93,4515	105,2573	107,8850	105,0886	93,2389
840,1310	99,8089	132,3565	132,5997	78,4522	78,4194	63,7272	63,1813	6,8426	,2578
85,0430	143,1457	97,5208	951,6724	397,0543	143,0706	85,0139	91,7214	103,8487	107,7883
103,9497	91,4611	101,5094							

## LOAD STEP # 178

.085b

120,9916	113,1118	110,3323	117,2429	117,6949	122,5735	117,4555	112,8580	111,1286	110,3849
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(TIMES .09 to 1.60 OMITTED FOR BREVITY)

LOAD STEP # 235  
1.6396

199,1990	199,1767	199,3899	199,4138	199,1898	199,2231	199,2731	199,2459	199,3592	199,3408
199,2233	199,2696	199,3992	199,2815	199,5768	199,5366	199,2529	199,3743	199,3160	199,1617
199,0391	199,0262	199,1531	199,3343	199,1157	199,1802	199,7733	199,7768	199,1469	199,0959
199,0837	199,0469	199,1989	199,2063	199,0942	199,1463	187,6353	198,7202	219,4002	628,0044
219,5392	198,7267	187,6305	187,3172	184,9920	187,1432	184,9645	187,3390	182,0260	186,2585
219,3103	660,6964	219,4976	146,3189	162,0350	177,1490	184,8175	183,7141	184,7934	177,1376
603,0007	171,5631	164,0275	164,0527	175,1234	155,1226	149,3894	149,4134	202,8709	17,8896
172,6830	180,981	219,1403	658,1339	219,2917	181,0423	172,6964	172,2332	184,1969	183,6560
184,1616	172,2165	199,3842							

LOAD STEP # 237  
1.7396

194,5678	194,5922	194,6994	194,7195	194,6064	194,5741	194,5592	194,6337	194,6859	194,6770
194,6249	194,5498	191,7258	194,6851	194,9145	194,8802	194,6662	194,7167	194,6242	194,5569
194,3868	194,3783	194,5514	194,6263	194,4753	194,5605	195,0882	195,0948	194,5270	194,4434
194,4419	194,4710	194,4824	194,4917	194,5143	194,4849	183,2427	194,1519	214,6607	616,5104
214,7907	194,1517	183,2437	182,9614	180,6381	182,6992	180,6263	182,9738	177,7610	182,0066
214,5493	608,8604	214,6938	182,0654	177,7773	172,9746	180,4659	179,3719	180,4520	172,9612
591,7716	167,4742	160,1097	160,1448	151,3414	151,3302	145,5803	145,5888	197,5430	18,8254
168,6195	176,8416	214,4088	646,2869	214,5563	176,9007	168,6390	168,1589	179,8521	179,3048
179,8153	168,1394	194,7318							

LOAD STEP # 239  
1.8396

190,5576	190,6035	190,6605	190,6791	190,6183	190,5641	190,5474	190,6349	190,6759	190,6658
190,6183	190,5327	190,6709	190,6793	190,9157	190,9052	190,6725	190,6518	190,5923	190,5515
190,3902	190,3923	190,5488	190,5981	190,4703	190,5391	191,0455	191,0670	190,5075	190,4239
190,4639	190,4887	190,4356	190,4462	190,5120	190,4944	179,4460	190,1737	210,5070	605,8706
210,6460	190,1671	179,4494	179,1997	176,8834	178,8812	176,8660	179,1968	174,0569	178,2901
210,3826	637,8593	210,5241	178,3471	174,7553	169,3589	176,7122	175,6283	176,7011	169,3460
581,4366	163,9346	156,6997	156,7355	148,0646	148,0590	142,2730	142,2805	192,8776	19,7363
165,0781	173,2187	210,2468	635,3059	210,3865	173,2777	165,0983	164,6283	176,1108	175,5598
176,0218	164,6137	190,7207							

LOAD STEP # 241  
1.9396

186,9339	186,9608	187,0152	187,0315	186,9691	186,9530	186,9465	186,9893	187,0825	187,0773
186,9740	186,9207	187,0329	187,0234	187,2946	187,2873	187,0290	187,0276	186,9694	186,8963
186,4029	186,8049	186,9002	186,9816	186,8386	186,8644	187,3672	187,3855	186,8330	186,7919
186,8524	186,8535	186,8249	186,8275	186,8698	186,8876	176,0045	186,5203	206,5566	594,9494
206,6873	186,5093	176,0031	175,7661	173,4910	175,4578	173,4599	175,7639	170,6922	174,8513
206,4285	626,5342	206,5670	174,9073	170,7096	166,0814	173,3204	172,2538	173,3081	166,0698
571,0290	160,7360	153,5769	153,6033	145,0979	145,1009	139,2727	139,2883	188,8115	20,8241
161,8540	169,8637	206,2787	623,9857	206,4218	169,9214	161,8760	161,4295	172,7266	172,1863
172,7121	161,4167	185,0967							

LOAD STEP # 243  
1.9996

184,8974	184,9331	184,9407	184,9667	184,9603	184,9134	184,8630	184,8629	184,8179	184,8100
184,8403	184,8408	184,7491	184,8137	185,0916	185,0821	184,8328	184,7836	184,8324	184,8439
184,8143	184,8071	184,8176	184,7911	184,8019	184,8481	185,2920	185,3206	184,8634	184,8275
184,7796	184,7453	184,6249	184,6187	184,7315	184,7543	174,0903	184,3955	204,1715	589,1640
204,3017	184,4143	174,1107	173,8022	171,4478	173,3277	171,4085	173,7802	168,7986	172,9653
205,2622	619,9003	204,3748	172,9963	168,8051	164,1873	171,3657	170,1949	171,2769	164,1793
565,7987	158,7724	151,6686	151,6794	143,3994	143,3972	137,5729	137,5581	186,5523	21,1461
160,0253	168,0346	204,3505	618,3976	204,4558	168,0670	160,0236	159,5997	170,7666	170,1442
170,7252	159,5976	184,8054							

EOS PRESSURE	TIME	ISTONY NUMBER	P	TIME	SEC TIME	PSIG PRESSURE
0.00000	0.0	.00500	.0	.01000	.0	.01500
.02000	.0	.02500	.1	.03000	.9	.03500
.04000	12.4	.04500	25.3	.05000	36.7	.05500
.06060	71.3	.06500	87.8	.07060	94.5	.07560
.08060	101.0	.08500	110.5	.09060	123.3	.09560
.10960	161.2	.12960	167.6	.14960	191.8	.16960
.18960	211.2	.21960	215.8	.26460	238.1	.31460
.36460	244.5	.43960	257.0	.53960	264.4	.63960
.73960	260.6	.81960	250.8	.93960	243.2	1.03960
1.13960	228.9	1.21960	223.1	1.33960	217.0	1.43960
1.53960	204.7	1.63960	199.3	1.73960	194.6	1.83960

SECTION 3: EOS-SNAT  
ELEMENT PRESSURE  
TIME HISTORIES

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.73960	240.6	1.23960	25.1	.93960	243.2	1.03960	235.4
1.13960	228.9	1.23960	223.1	1.33960	217.0	1.43960	210.8
1.53960	204.7	1.63960	199.3	1.73960	194.6	1.83960	190.6
1.93960	187.0	1.99960	184.8				

EDS PRESSURE TIME HISTORY NUMBER 2

2 46							
0.00000	0.0	.00500	.0	.01000	.0	.01500	.0
.02000	.1	.02500	.5	.03000	2.7	.03500	7.9
.04000	15.0	.04500	25.5	.05000	34.9	.05500	56.5
.06000	69.0	.06500	78.6	.07000	88.3	.07500	97.0
.08000	103.6	.08500	112.0	.09000	125.1	.09500	139.9
.10960	161.0	.12960	162.8	.14960	184.4	.16960	196.2
.18960	210.9	.21460	218.1	.26460	232.8	.31460	245.0
.36460	250.6	.43960	255.5	.53960	266.7	.63960	267.4
.73960	260.1	.83960	251.7	.93960	243.4	1.03960	235.4
1.13960	228.9	1.23960	223.0	1.33960	216.9	1.43960	210.6
1.53960	204.6	1.63960	199.2	1.73960	194.6	1.83960	190.6
1.93960	187.0	1.99960	184.8				

EDS PRESSURE TIME HISTORY NUMBER 3

3 46							
0.00000	0.0	.00500	.0	.01000	.0	.01500	.0
.02000	.3	.02500	1.9	.03000	6.1	.03500	12.2
.04000	19.0	.04500	29.7	.05000	43.9	.05500	57.6
.06000	65.9	.06500	71.0	.07000	79.6	.07500	93.2
.08000	106.6	.08500	118.1	.09000	128.3	.09500	135.4
.10960	163.5	.12960	171.4	.14960	190.7	.16960	196.7
.18960	214.6	.21460	218.8	.26460	235.3	.31460	245.7
.36460	247.6	.43960	258.5	.53960	265.4	.63960	264.3
.73960	259.7	.83960	250.9	.93960	243.3	1.03960	235.2
1.13960	229.0	1.23960	223.1	1.33960	217.1	1.43960	210.7
1.53960	204.7	1.63960	199.3	1.73960	194.6	1.83960	190.5
1.93960	186.9	1.99960	184.8				

EDS PRESSURE TIME HISTORY NUMBER 4

4 46							
0.00000	0.0	.00500	.0	.01000	.0	.01500	.1
.02000	1.2	.02500	4.5	.03000	10.1	.03500	16.5
.04000	25.0	.04500	37.0	.05000	47.5	.05500	54.6
.06000	60.6	.06500	67.9	.07000	75.7	.07500	86.5
.08000	103.7	.08500	120.5	.09000	129.2	.09500	137.5
.10960	150.8	.12960	172.2	.14960	192.3	.16960	192.9
.18960	210.3	.21460	222.9	.26460	236.1	.31460	241.4
.36460	249.3	.43960	263.5	.53960	266.5	.63960	263.6
.73960	259.3	.83960	251.4	.93960	242.6	1.03960	235.5
1.13960	228.9	1.23960	223.1	1.33960	216.9	1.43960	210.7
1.53960	204.7	1.63960	199.2	1.73960	194.6	1.83960	190.6
1.93960	186.9	1.99960	184.9				

EDS PRESSURE TIME HISTORY NUMBER 5

5 46							
0.00000	0.0	.00500	.0	.01000	.0	.01500	.6
.02000	2.8	.02500	7.4	.03000	13.5	.03500	21.7
.04000	31.5	.04500	42	.05000	48.6	.05500	52.3
.06000	57.1	.06500	64.7	.07000	74.0	.07500	85.0
.08000	96.8	.08500	112.2	.09000	134.3	.09500	151.1
.10960	144.9	.12960	176.5	.14960	190.3	.16960	190.2
.18960	206.4	.21460	217.1	.26460	233.0	.31460	242.8
.36460	248.3	.43960	270.6	.53960	269.6	.63960	264.0
.73960	260.7	.83960	251.5	.93960	243.1	1.03960	235.4
1.13960	229.1	1.23960	222.9	1.33960	216.9	1.43960	210.5
1.53960	204.5	1.63960	199.2	1.73960	194.6	1.83960	190.6
1.93960	187.0	1.99960	184.9				

EDS PRESSURE TIME HISTORY NUMBER 6

6 46							
0.00000	0.0	.00500	.0	.01000	.2	.01500	1.4
.02000	4.4	.02500	9.3	.03000	16.3	.03500	25.4

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0.00000	0.0	.00500	.0	.01000	.2	.01500	1.4
.02000	4.4	.02500	9.3	.03000	14.3	.03500	25.4
.04000	34.4	.04500	42.7	.05000	49.3	.05500	53.0
.06000	57.6	.06500	62.5	.07000	67.7	.07500	74.6
.08000	95.9	.08500	117.6	.09000	135.3	.09500	146.1
.10000	115.0	.10500	170.0	.11000	166.7	.11500	196.9
.12000	214.0	.12500	222.3	.13000	231.4	.13500	244.6
.14000	251.0	.14500	267.1	.15000	268.4	.15500	263.7
.16000	256.9	.16500	251.0	.17000	242.9	.17500	235.7
.18000	229.0	.18500	223.2	.19000	217.1	.19500	210.9
.20000	204.9	.20500	199.4	.21000	194.1	.21500	190.7
.22000	187.1	.22500	185.0	.23000		.23500	

EDS PRESSURE TIME HISTORY NUMBER 7

0.00000	0.0	.00500	.0	.01000	.0	.01500	.0
.02000	0.0	.02500	.4	.03000	2.3	.03500	6.6
.04000	12.2	.04500	20.9	.05000	34.5	.05500	51.5
.06000	67.0	.06500	82.3	.07000	95.0	.07500	100.6
.08000	103.9	.08500	112.7	.09000	127.1	.09500	139.5
.10000	150.1	.10500	168.7	.11000	170.2	.11500	211.7
.12000	211.3	.12500	215.3	.13000	234.9	.13500	246.1
.14000	245.1	.14500	261.8	.15000	259.0	.15500	260.2
.16000	260.0	.16500	251.0	.17000	243.1	.17500	235.1
.18000	228.8	.18500	222.9	.19000	216.8	.19500	210.6
.20000	204.5	.20500	199.1	.21000	194.4	.21500	190.4
.22000	186.8	.22500	184.8	.23000		.23500	

EDS PRESSURE TIME HISTORY NUMBER 8

0.00000	0.0	.00500	.0	.01000	.0	.01500	.0
.02000	.3	.02500	1.8	.03000	5.1	.03500	8.6
.04000	13.5	.04500	25.5	.05000	39.6	.05500	53.0
.06000	64.7	.06500	76.3	.07000	86.7	.07500	99.9
.08000	107.0	.08500	116.3	.09000	128.8	.09500	133.8
.10000	157.2	.10500	167.0	.11000	160.2	.11500	201.0
.12000	203.6	.12500	216.4	.13000	230.7	.13500	244.6
.14000	251.1	.14500	261.7	.15000	265.0	.15500	260.3
.16000	260.6	.16500	251.4	.17000	240.0	.17500	235.6
.18000	229.5	.18500	223.0	.19000	216.8	.19500	210.5
.20000	186.9	.20500	184.8	.21000	194.5	.21500	190.5

EDS PRESSURE TIME HISTORY NUMBER 9

0.00000	0.0	.00500	.0	.01000	.0	.01500	.2
.02000	1.4	.02500	4.4	.03000	6.5	.03500	11.7
.04000	18.8	.04500	33.0	.05000	45.5	.05500	53.1
.06000	60.2	.06500	70.9	.07000	82.2	.07500	94.0
.08000	110.7	.08500	124.5	.09000	123.7	.09500	121.4
.10000	153.2	.10500	164.5	.11000	162.4	.11500	205.9
.12000	209.6	.12500	217.7	.13000	232.5	.13500	245.3
.14000	248.4	.14500	262.0	.15000	263.9	.15500	265.4
.16000	25.0	.16500	250.5	.17000	243.2	.17500	235.4
.18000	229.1	.18500	223.0	.19000	217.1	.19500	210.6
.20000	204.7	.20500	199.3	.21000	194.6	.21500	190.6
.22000	187.0	.22500	184.8	.23000		.23500	

EDS PRESSURE TIME HISTORY NUMBER 10

0.00000	0.0	.00500	.0	.01000	.1	.01500	1.1
.02000	4.4	.02500	4.4	.03000	12.1	.03500	17.5
.04000	28.7	.04500	41.9	.05000	48.1	.05500	52.0
.06000	58.3	.06500	66.9	.07000	77.2	.07500	94.3
.08000	112.3	.08500	114.4	.09000	115.9	.09500	128.4
.10000	109.2	.10500	164.4	.11000	189.4	.11500	202.8
.12000	206.0	.12500	222.4	.13000	235.3	.13500	247.0
.14000	248.9	.14500	241.9	.15000	265.6	.15500	263.4
.16000	163.0	.16500	163.0	.17000	163.0	.17500	163.0

10960	199.2	12960	165.4	10960	189.9	10960	202.8
18960	206.0	21960	226.0	20960	235.3	22960	247.0
30960	268.0	33960	263.0	35960	265.0	38960	263.0
40960	259.7	43960	251.3	45960	292.7	48960	235.0
50960	229.0	52960	223.2	54960	217.0	57960	210.8
60960	202.8	63960	190.3	65960	194.7	68960	190.0
70960	187.0	73960	184.8				

EDS PRESSURE TIME HISTORY NUMBER 11

0.00000	0.0	00500	0	01000	0.6	01500	3.5
02000	8.5	02500	13.9	03000	20.5	03500	32.8
04000	48.3	04500	50.5	05000	61.5	05500	63.8
06000	69.9	06500	78.8	07000	67.7	07500	97.0
08000	96.7	08500	105.9	09000	123.2	09500	125.1
10960	160.7	10960	160.1	10960	186.9	10960	176.0
16960	203.9	21960	219.6	26960	233.8	31960	249.0
36960	247.4	41960	264.7	46960	264.5	51960	263.9
73960	261.4	83960	251.7	93960	243.0	103960	235.0
113960	229.1	123960	222.0	133960	216.9	143960	210.7
153960	204.6	163960	190.3	173960	194.7		190.7
193960	187.0	199960	184.8				

EDS PRESSURE TIME HISTORY NUMBER 12

0.00000	0.0	00500	3	01000	2.4	01500	7.3
02000	14.2	02500	22.1	03000	33.6	03500	46.2
04000	52.7	04500	34.7	05000	54.8	05500	65.1
06000	72.7	06500	70.4	07000	85.0	07500	92.2
08000	108.6	08500	118.1	09000	110.9	09500	110.2
10960	165.8	10960	160.1	10960	185.3	10960	166.7
16960	209.3	21960	223.1	26960	235.8	31960	245.6
36960	256.0	41960	261.0	46960	271.0	51960	263.4
73960	250.6	83960	251.1	93960	243.3	103960	236.0
113960	229.4	123960	223.5	133960	217.4	143960	211.1
153960	207.0	163960	192.6	173960	194.9		190.9
193960	187.3	199960	185.1				

EDS PRESSURE TIME HISTORY NUMBER 13

0.00000	0.0	00500	0	01000	0	01500	0
02000	1	02500	8	03000	3.2	03500	7.2
04000	12.0	04500	19.3	05000	29.7	05500	46.3
06000	65.0	06500	79.3	07000	89.5	07500	100.0
08000	110.5	08500	118.0	09000	123.8	09500	130.6
10960	143.2	10960	149.0	10960	176.6	10960	196.8
16960	201.0	21960	217.1	26960	238.9	31960	241.8
36960	245.2	41960	256.5	46960	259.1	51960	260.7
73960	258.0	83960	248.0	93960	240.4	103960	232.7
113960	226.3	123960	220.5	133960	214.5	143960	208.3
153960	202.4	163960	197.0	173960	192.3		184.3
193960	184.7	199960	182.5				

EDS PRESSURE TIME HISTORY NUMBER 14

0.00000	0.0	00500	0	01000	0	01500	1
02000	13.7	02500	2.8	03000	5.7	03500	9.0
04000	62.4	04500	23.2	05000	35.3	05500	49.6
06000	111.1	06500	74.4	07000	86.0	07500	98.9
08000	188.7	08500	121.4	09000	125.2	09500	128.3
10960	202.5	10960	175.0	10960	181.3	10960	201.5
16960	245.4	21960	211.0	26960	230.6	31960	245.3
36960	248.0	41960	262.4	46960	263.4	51960	282.6
73960	256.0	83960	248.7	93960	241.2	103960	233.1
113960	226.0	123960	220.6	133960	214.7	143960	204.0
153960	202.0	163960	197.3	173960	192.7		184.7
193960	185.1	199960	183.0				

EDS PRESSURE TIME HISTORY NUMBER 15

1.9396 185.1 1.9996 183.0  
 EOS PRESSURE TIME HISTORY NUMBER 15  
 15 46  
 0.0000 0.0 .0500 7.0  
 .0200 2.8 .0250 6.1  
 .0400 20.1 .0560 33.5  
 .0600 59.4 .0650 65.3  
 .0800 113.7 .0850 122.5  
 .1000 146.6 .1200 175.4  
 .1800 204.3 .2100 220.3  
 .2400 246.5 .3300 264.1  
 .3900 288.1 .4300 241.2  
 1.1396 227.0 1.2396 221.1  
 1.5396 202.8 1.6396 187.5  
 1.9396 185.3 1.9996 183.2

.01000 9.2  
 .03000 45.6  
 .05060 80.5  
 .07060 119.3  
 .09060 179.7  
 .14960 234.1  
 .26460 264.1  
 .33960 264.1  
 .43960 215.1  
 1.43960 192.9  
 1.83960 184.9

EOS PRESSURE TIME HISTORY NUMBER 16  
 16 46  
 0.0000 0.0 .0500 11.4  
 .0200 7.0 .0250 48.6  
 .0400 30.6 .0560 71.2  
 .0600 65.5 .0850 105.4  
 .0800 114.3 .1200 172.8  
 .1000 149.4 .2100 220.0  
 .1800 198.1 .3300 254.4  
 .2400 248.6 .3900 289.9  
 .4300 258.2  
 1.1396 227.0 1.2396 222.1  
 1.5396 203.7 1.6396 184.3  
 1.9396 186.0 1.9996 184.0

.01000 15.3  
 .03000 56.6  
 .05060 82.0  
 .07060 115.5  
 .09060 163.2  
 .14960 236.0  
 .26460 265.6  
 .33960 265.6  
 .43960 241.6  
 1.43960 215.9  
 1.83960 193.7

EOS PRESSURE TIME HISTORY NUMBER 17  
 17 46  
 0.0000 0.0 .0500 28.6  
 .0200 15.3 .0250 123.3  
 .0400 68.2 .0560 148.1  
 .0600 142.6 .0850 194.3  
 .0800 114.4 .1200 187.7  
 .1000 164.6 .2100 236.6  
 .1800 189.6 .3300 268.7  
 .2400 262.0 .3900 253.1  
 .4300 253.1  
 1.1396 230.5 1.2396 224.5  
 1.5396 206.0 1.6396 200.6  
 1.9396 188.2 1.9996 186.2

.01000 43.0  
 .03000 133.6  
 .05060 152.9  
 .07060 126.6  
 .09060 206.5  
 .14960 251.7  
 .26460 273.3  
 .33960 248.3  
 .43960 248.3  
 1.43960 196.6  
 1.83960 191.9

EOS PRESSURE TIME HISTORY NUMBER 18  
 18 46  
 0.0000 0.0 .0500 4.3  
 .0200 40.3 .0250 97.3  
 .0400 195.3 .0560 196.0  
 .0600 211.5 .0850 220.3  
 .0800 220.0 .0900 191.9  
 .1000 220.0 .1200 261.3  
 .1800 248.7 .2100 309.8  
 .1900 290.4 .3300 332.3  
 .2400 333.3 .3900 333.4  
 .4300 315.4  
 1.1396 277.2 1.2396 270.1  
 1.5396 248.3 1.6396 242.1  
 1.9396 227.7 1.9996 225.3

.01000 16.6  
 .03000 140.2  
 .05060 197.5  
 .07060 226.4  
 .09060 191.9  
 .14960 283.5  
 .26460 316.2  
 .33960 331.7  
 .43960 293.0  
 1.43960 262.6  
 1.83960 236.6

EOS PRESSURE TIME HISTORY NUMBER 19  
 19 46  
 0.0000 0.0 .0500 1.1  
 .0200 11.3 .0250 17.1  
 .0400 56.5 .0560 70.4  
 .0600 108.1 .0850 112.9

.01000 3.5  
 .03000 25.2  
 .05060 62.9  
 .07060 117.9  
 .09060 117.9

EDS PRESSURE TIME HISTORY NUMBER 20

0.0000	11.3	0.0500	17.1	0.0500	25.2	0.0500	39.1
0.0000	50.5	0.0500	70.9	0.0700	82.9	0.0750	96.0
0.0000	108.7	0.0500	112.9	0.0900	117.0	0.0950	122.0
0.0000	137.6	0.0700	150.4	0.0900	160.9	0.1000	161.9
0.0000	191.5	0.1000	203.1	0.1000	221.5	0.1000	227.1
0.0000	231.0	0.1000	240.1	0.1000	266.7	0.1000	246.9
0.0000	210.1	0.1000	255.3	0.1000	287.3	0.1000	219.8
0.0000	213.7	0.1000	200.2	0.1000	202.4	0.1000	196.5
0.0000	190.8	0.1000	185.6	0.1000	181.4	0.1000	177.6
0.0000	174.2	0.1000	172.1	0.1000	173.960	0.1000	173.960

EDS PRESSURE TIME HISTORY NUMBER 21

0.0000	0.0	0.0500	0	0.1000	0	0.1500	1
0.0000	3.7	0.0500	2.4	0.3000	4.9	0.3500	8.4
0.0000	13.0	0.0500	17.3	0.5000	26.7	0.5500	43.1
0.0000	57.0	0.0500	67.9	0.7000	62.6	0.7500	96.6
0.0000	106.5	0.0500	115.0	0.9000	120.0	0.9500	120.6
0.0000	137.8	0.0500	158.3	1.0000	166.9	1.0000	184.8
0.0000	193.0	0.0500	202.6	1.0000	220.1	0.1000	228.4
0.0000	230.3	0.0500	247.6	0.1000	248.1	0.1000	247.6
0.0000	243.6	0.0500	235.2	0.1000	221.5	0.1000	219.9
0.0000	213.6	0.0500	208.2	0.1000	202.4	0.1000	196.5
0.0000	190.9	0.0500	185.8	0.1000	181.5	0.1000	177.7
0.0000	174.3	0.0500	172.2	0.1000	173.960	0.1000	173.960

EDS PRESSURE TIME HISTORY NUMBER 22

0.0000	0.0	0.0500	0	0.1000	1	0.1500	1.2
0.0000	3.7	0.0500	6.5	0.3000	10.1	0.3500	15.2
0.0000	22.7	0.0500	36.0	0.5000	48.0	0.5500	58.8
0.0000	64.9	0.0500	70.1	0.7000	78.9	0.7500	88.7
0.0000	106.5	0.0500	121.5	0.9000	122.1	0.9500	118.2
0.0000	138.6	0.0500	149.3	1.0000	149.6	1.0000	167.5
0.0000	193.6	0.0500	209.1	0.1000	248.0	0.1000	229.2
0.0000	233.0	0.0500	282.0	0.1000	252.4	0.1000	249.9
0.0000	245.3	0.0500	237.1	0.1000	229.3	0.1000	219.9
0.0000	215.6	0.0500	210.1	0.1000	204.3	0.1000	198.4
0.0000	192.6	0.0500	187.5	0.1000	183.1	0.1000	179.4
0.0000	175.9	0.0500	174.0	0.1000	173.960	0.1000	173.960

EDS PRESSURE TIME HISTORY NUMBER 23

0.0000	0.0	0.0500	0	0.1000	0	0.1500	3.4
0.0000	6.8	0.0500	13.4	0.3000	23.2	0.3500	33.9
0.0000	46.3	0.0500	60.1	0.5000	63.5	0.5500	91.5
0.0000	97.0	0.0500	101.5	0.7000	105.9	0.7500	111.1
0.0000	116.1	0.0500	121.2	0.9000	123.6	0.9500	131.7
0.0000	138.7	0.0500	149.6	1.0000	172.6	1.0000	160.4
0.0000	196.3	0.0500	215.5	0.1000	227.4	0.1000	232.4
0.0000	242.4	0.0500	250.9	0.1000	258.9	0.1000	254.7
0.0000	250.1	0.0500	242.1	0.1000	233.7	0.1000	226.6
0.0000	220.3	0.0500	214.9	0.1000	208.6	0.1000	204.5
0.0000	196.7	0.0500	191.5	0.1000	187.1	0.1000	183.2
0.0000	179.7	0.0500	177.7	0.1000	173.960	0.1000	173.960

EDS PRESSURE TIME HISTORY NUMBER 24

0.0000	0.0	0.0500	1.3	0.1000	8.5	0.1500	20.2
0.0000	36.7	0.0500	70.9	0.3000	145.0	0.3500	159.5
0.0000	107.6	0.0500	233.0	0.5000	257.8	0.5500	275.7
0.0000	209.0	0.0500	302.3	0.7000	310.7	0.7500	313.7
0.0000	306.9	0.0500	305.5	0.9000	311.2	0.9500	314.9
0.0000	325.2	0.0500	337.0	1.0000	341.8	1.0000	346.4
0.0000	350.4	0.0500	358.1	0.1000	341.8	0.1000	356.0
0.0000	335.5	0.0500	288.3	0.1000	289.1	0.1000	283.5
0.0000	276.6	0.0500	266.4	0.1000	257.2	0.1000	249.8
0.0000	243.3	0.0500	237.1	0.1000	230.6	0.1000	223.4
0.0000	217.6	0.0500	212.1	0.1000	207.4	0.1000	203.3



EDS PRESSURE TIME HISTORY NUMBER 24

73960	276.0	.43960	260.4	.93960	257.2	1.03960	249.4
1.13960	243.3	1.23960	237.1	1.33960	230.6	1.43960	223.9
1.53960	217.0	1.63960	212.1	1.73960	207.4	1.83960	203.3
1.93960	199.5	1.99960	197.2				

EDS PRESSURE TIME HISTORY NUMBER 24

24	46	0.0	.05000	.01000	49.4	.01500	92.5
0.00000	161.7	.02500	.251.6	.03000	353.6	.03500	440.2
.02000	.04000	.04560	513.3	.05060	523.3	.05560	531.0
.06000	541.4	.06560	555.0	.07060	568.5	.07560	578.9
.08000	586.4	.08560	591.5	.09060	594.4	.09560	595.4
.10960	600.1	.12960	603.6	.14960	609.0	.16960	608.3
.18960	608.0	.21960	611.2	.24960	606.6	.31460	566.7
.36460	561.2	.43960	530.8	.53960	510.6	.63960	502.2
.73960	482.4	.83960	455.6	.93960	438.9	1.03960	427.2
1.13960	417.6	1.23960	407.4	1.33960	396.5	1.43960	385.3
1.53960	375.5	1.63960	367.1	1.73960	360.0	1.83960	353.5
1.93960	347.0	1.99960	343.4				

EDS PRESSURE TIME HISTORY NUMBER 25

25	46	0.0	.00500	.01000	3.4	.01500	.0
0.00000	10.9	.02500	1.1	.03000	3.4	.03500	9.7
.02000	.04000	.04560	16.5	.05060	23.5	.05560	35.6
.06000	106.0	.06560	69.0	.07060	64.6	.07560	99.2
.08000	134.2	.08560	110.3	.09060	115.8	.09560	121.2
.10960	192.4	.12960	133.3	.14960	160.9	.16960	162.5
.18960	229.6	.21960	201.2	.24960	219.7	.31460	226.2
.36460	243.0	.43960	246.4	.53960	245.5	.63960	246.7
.73960	212.0	.83960	230.3	.93960	226.4	1.03960	219.0
1.13960	190.5	1.23960	207.3	1.33960	201.5	1.43960	195.7
1.53960	173.5	1.63960	185.0	1.73960	180.6	1.83960	176.9
1.93960		1.99960	171.4				

EDS PRESSURE TIME HISTORY NUMBER 26

26	46	0.0	.00500	.01000	3.8	.01500	.0
0.00000	11.3	.02500	1.4	.03000	3.8	.03500	7.2
.02000	.04000	.04560	16.7	.05060	23.5	.05560	36.5
.06000	103.9	.06560	69.1	.07060	64.3	.07560	98.4
.08000	133.0	.08560	110.3	.09060	116.4	.09560	121.3
.10960	192.1	.12960	133.5	.14960	166.3	.16960	162.8
.18960	229.0	.21960	201.0	.24960	219.3	.31460	226.1
.36460	242.5	.43960	246.4	.53960	245.7	.63960	246.4
.73960	212.5	.83960	233.9	.93960	226.0	1.03960	218.7
1.13960	189.7	1.23960	207.0	1.33960	201.2	1.43960	195.3
1.53960	173.2	1.63960	184.7	1.73960	180.3	1.83960	176.6
1.93960		1.99960	171.2				

EDS PRESSURE TIME HISTORY NUMBER 27

27	46	0.0	.00500	.01000	1.1	.01500	1.5
0.00000	4.5	.02500	7.6	.03000	11.6	.03500	17.9
.02000	.04000	.04560	42.5	.05060	56.0	.05560	64.5
.06000	64.7	.06560	72.2	.07060	76.1	.07560	82.7
.08000	101.1	.08560	120.9	.09060	121.6	.09560	118.3
.10960	126.5	.12960	159.5	.14960	161.6	.16960	164.4
.18960	190.0	.21960	205.4	.24960	217.6	.31460	224.6
.36460	229.5	.43960	249.6	.53960	246.2	.63960	246.4
.73960	241.7	.83960	233.6	.93960	225.4	1.03960	218.7
1.13960	212.5	1.23960	206.9	1.33960	201.1	1.43960	195.3
1.53960	189.6	1.63960	184.6	1.73960	180.3	1.83960	176.6
1.93960		1.99960	171.2				

EDS PRESSURE TIME HISTORY NUMBER 28

28	46	0.0	.00500	.01000	.3	.01500	2.0
0.00000	5.3	.02500	9.3	.03000	14.6	.03500	22.4
.02000							

28	46						
0.00000	0.0	.00500	.0	.01000	.3	.01500	2.0
.02000	5.3	.02500	9.3	.03000	14.6	.03500	22.4
.04000	33.2	.04500	50.2	.05000	64.5	.05500	72.9
.06000	76.2	.06500	79.6	.07000	82.5	.07500	88.0
.08000	104.5	.08500	121.7	.09000	122.1	.09500	119.3
.10000	127.9	.12000	160.6	.14000	165.9	.16000	145.2
.18000	190.7	.21000	200.6	.26000	218.6	.31000	225.5
.36000	230.9	.43000	250.8	.53000	250.0	.63000	247.4
.73000	242.7	.83000	234.6	.93000	226.7	1.03000	219.6
1.13000	213.4	1.23000	207.8	1.33000	202.0	1.43000	196.1
1.53000	190.5	1.63000	185.0	1.73000	181.1	1.83000	177.3
1.93000	173.9	1.99000	172.0				

EDS PRESSURE TIME HISTORY NUMBER 29

29	46						
0.00000	0.0	.00500	1.7	.01000	10.9	.01500	25.6
.02000	52.5	.02500	44.3	.03000	145.3	.03500	199.0
.04000	249.3	.04500	249.3	.05000	314.7	.05500	335.0
.06000	352.1	.06500	363.4	.07000	371.3	.07500	377.6
.08000	381.6	.08500	384.6	.09000	388.1	.09500	391.6
.10000	398.3	.12000	404.3	.14000	408.5	.16000	411.3
.18000	411.7	.21000	413.9	.26000	413.3	.31000	404.7
.36000	380.7	.43000	299.4	.53000	296.8	.63000	291.8
.73000	243.5	.83000	272.4	.93000	263.0	1.03000	255.5
1.13000	249.0	1.23000	202.6	1.33000	236.0	1.43000	229.1
1.53000	222.8	1.63000	217.2	1.73000	212.5	1.83000	208.4
1.93000	204.4	1.99000	202.2				

EDS PRESSURE TIME HISTORY NUMBER 30

PTM NEXT TO HOT LEG BREAK

30	46						
0.00000	0.0	.00500	6.9	.01000	27.0	.01500	56.4
.02000	105.2	.02500	172.9	.03000	247.8	.03500	320.7
.04000	371.7	.04500	409.4	.05000	434.3	.05500	454.1
.06000	468.4	.06500	477.7	.07000	485.1	.07500	492.7
.08000	499.9	.08500	504.8	.09000	507.6	.09500	509.7
.10000	515.2	.12000	519.0	.14000	523.7	.16000	524.7
.18000	524.5	.21000	526.4	.26000	523.1	.31000	509.6
.36000	486.2	.43000	425.2	.53000	410.5	.63000	403.6
.73000	390.3	.83000	370.0	.93000	356.8	1.03000	347.1
1.13000	339.0	1.23000	350.6	1.33000	321.7	1.43000	312.6
1.53000	304.4	1.63000	297.4	1.73000	291.3	1.83000	286.0
1.93000	280.7	1.99000	277.7				

EDS PRESSURE TIME HISTORY NUMBER 31

31	46						
0.00000	0.0	.00500	.0	.01000	.0	.01500	.0
.02000	.1	.02500	.8	.03000	2.8	.03500	6.0
.04000	10.0	.04500	15.6	.05000	21.7	.05500	31.5
.06000	47.9	.06500	67.0	.07000	86.3	.07500	100.8
.08000	106.1	.08500	108.3	.09000	113.1	.09500	119.1
.10000	131.7	.12000	150.0	.14000	167.8	.16000	182.4
.18000	193.3	.21000	199.4	.26000	218.1	.31000	225.0
.36000	228.6	.43000	246.4	.53000	243.7	.63000	246.5
.73000	242.3	.83000	233.5	.93000	225.8	1.03000	218.5
1.13000	212.2	1.23000	206.7	1.33000	200.9	1.43000	195.1
1.53000	189.4	1.63000	184.4	1.73000	180.3	1.83000	176.3
1.93000	172.9	1.99000	170.9				

EDS PRESSURE TIME HISTORY NUMBER 32

32	46						
0.00000	0.0	.00500	.0	.01000	.0	.01500	.1
.02000	.5	.02500	1.9	.03000	4.3	.03500	7.6
.04000	10.9	.04500	16.5	.05000	23.0	.05500	34.0
.06000	51.3	.06500	68.9	.07000	84.1	.07500	95.6
.08000	100.9	.08500	107.6	.09000	115.5	.09500	120.5
.10000	124.3	.12000	149.3	.14000	164.0	.16000	181.3
.18000	190.1	.21000	198.0	.26000	215.8	.31000	222.3
.36000	225.7	.43000	244.4	.53000	242.3	.63000	243.7

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.10960	124.3	.12960	149.3	.14960	164.0	.16960	181.3
.18960	190.1	.21460	198.0	.26460	215.4	.31460	222.3
.36460	225.7	.43960	244.4	.53960	242.3	.63960	243.7
.73960	239.2	.83960	230.7	.93960	222.9	1.03960	215.6
1.13960	209.6	1.23960	200.1	1.33960	194.4	1.43960	192.6
1.53960	187.1	1.63960	182.1	1.73960	177.8	1.83960	174.1
1.93960	170.8	1.99960	168.8				

EDS PRESSURE TIME HISTORY NUMBER 33

33 46							
0.00000	0.0	.00500	.0	.01000	.1	.01500	1.0
.02000	3.4	.02500	6.4	.03000	10.0	.03500	15.8
.04000	23.5	.04500	33.4	.05000	47.0	.05500	57.8
.06000	64.5	.06500	71.1	.07000	76.2	.07500	81.3
.08000	94.2	.08500	115.9	.09000	120.7	.09500	118.7
.10960	117.8	.12960	151.6	.14960	160.9	.16960	140.9
.18960	186.3	.21460	198.0	.26460	212.0	.31460	219.7
.36460	223.6	.43960	244.4	.53960	243.0	.63960	241.5
.73960	236.5	.83960	228.4	.93960	220.7	1.03960	213.7
1.13960	207.6	1.23960	202.1	1.33960	196.4	1.43960	190.7
1.53960	185.2	1.63960	180.3	1.73960	176.1	1.83960	172.4
1.93960	169.1	1.99960	167.2				

EDS PRESSURE TIME HISTORY NUMBER 34

34 46							
0.00000	0.0	.00500	.0	.01000	.6	.01500	3.3
.02000	7.4	.02500	14.8	.03000	23.6	.03500	36.6
.04000	53.0	.04500	69.5	.05000	87.0	.05500	99.5
.06000	103.3	.06500	105.5	.07000	105.7	.07500	106.4
.08000	114.3	.08500	123.9	.09000	123.5	.09500	122.1
.10960	125.8	.12960	157.6	.14960	166.6	.16960	164.8
.18960	190.2	.21460	202.0	.26460	215.0	.31460	222.7
.36460	226.6	.43960	250.9	.53960	249.1	.63960	245.9
.73960	240.5	.83960	232.2	.93960	224.3	1.03960	217.4
1.13960	211.3	1.23960	205.7	1.33960	194.9	1.43960	194.0
1.53960	188.5	1.63960	183.5	1.73960	179.3	1.83960	175.5
1.93960	172.2	1.99960	170.3				

EDS PRESSURE TIME HISTORY NUMBER 35

35 46							
0.00000	0.0	.00500	1.2	.01000	8.1	.01500	19.7
.02000	42.7	.02500	76.9	.03000	115.9	.03500	160.8
.04000	206.1	.04500	242.2	.05000	265.0	.05500	285.2
.06000	298.6	.06500	304.3	.07000	307.5	.07500	311.6
.08000	313.9	.08500	309.8	.09000	310.5	.09500	314.3
.10960	320.1	.12960	332.2	.14960	339.1	.16960	345.0
.18960	347.3	.21460	351.3	.26460	354.9	.31460	351.3
.36460	344.5	.43960	287.8	.53960	283.9	.63960	279.2
.73960	271.5	.83960	261.1	.93960	252.0	1.03960	244.7
1.13960	238.4	1.23960	232.1	1.33960	225.8	1.43960	219.2
1.53960	213.1	1.63960	207.7	1.73960	203.1	1.83960	199.1
1.93960	195.3	1.99960	193.2				

EDS PRESSURE TIME HISTORY NUMBER 36

36 46							
0.00000	0.0	.00500	11.1	.01000	39.1	.01500	76.9
.02000	144.0	.02500	235.0	.03000	324.0	.03500	407.2
.04000	662.2	.04500	509.3	.05000	547.1	.05500	575.9
.06000	588.5	.06500	588.3	.07000	587.3	.07500	593.1
.08000	602.0	.08500	607.5	.09000	606.8	.09500	609.2
.10960	614.4	.12960	617.8	.14960	623.5	.16960	624.6
.18960	623.5	.21460	626.2	.26460	621.5	.31460	605.4
.36460	592.9	.43960	553.5	.53960	525.0	.63960	516.4
.73960	496.2	.83960	468.9	.93960	451.8	1.03960	440.1
1.13960	430.3	1.23960	419.9	1.33960	408.8	1.43960	397.4
1.53960	387.4	1.63960	378.9	1.73960	371.6	1.83960	365.0
1.93960	358.3	1.99960	354.6				

EDS PRESSURE TIME HISTORY NUMBER 37

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1.93960	358.3	1.99960	374.6				
EDS PRESSURE TIME HISTORY NUMBER 37							
37	46						
0.00000	77.0	.00500	77.0	.01000	77.0	.01500	77.0
.02000	77.1	.02500	77.1	.03000	77.2	.03500	77.5
.04000	77.8	.04500	77.9	.05000	78.0	.05500	78.8
.06000	79.6	.06500	79.5	.07000	79.6	.07500	80.6
.08000	106.1	.08500	107.7	.09000	110.3	.09500	118.5
.10960	132.2	.12960	148.0	.14960	168.1	.16960	183.5
.18960	193.7	.21960	198.0	.26460	217.0	.31460	224.2
.36460	229.0	.41960	245.9	.53960	242.2	.63960	246.3
.73960	241.4	.83960	232.9	.93960	225.4	1.03960	218.1
1.13960	211.7	1.23960	206.2	1.33960	200.5	1.43960	194.7
1.53960	189.0	1.63960	184.0	1.73960	179.6	1.83960	175.9
1.93960	172.5	1.99960	170.5				

EDS PRESSURE TIME HISTORY NUMBER 38							
38	46						
0.00000	0.1	.00500	0.1	.01000	0.1	.01500	0.1
.02000	0.6	.02500	2.0	.03000	4.5	.03500	7.9
.04000	11.0	.04500	15.8	.05000	23.1	.05500	33.1
.06000	49.4	.06500	70.1	.07000	84.4	.07500	91.6
.08000	98.1	.08500	104.2	.09000	113.0	.09500	121.8
.10960	118.8	.12960	144.8	.14960	161.2	.16960	179.6
.18960	187.0	.21960	194.5	.26460	211.9	.31460	218.0
.36460	221.7	.43960	241.6	.53960	238.1	.63960	239.9
.73960	235.3	.83960	226.7	.93960	219.2	1.03960	212.1
1.13960	206.0	1.23960	200.6	1.33960	194.9	1.43960	189.3
1.53960	183.8	1.63960	178.9	1.73960	174.7	1.83960	171.0
1.93960	167.7	1.99960	165.8				

EDS PRESSURE TIME HISTORY NUMBER 39							
39	46						
0.00000	0.0	.00500	0.0	.01000	0.0	.01500	0.0
.02000	2.6	.02500	5.6	.03000	9.3	.03500	14.0
.04000	20.5	.04500	29.9	.05000	40.3	.05500	52.8
.06000	64.1	.06500	70.5	.07000	74.9	.07500	81.2
.08000	89.7	.08500	104.8	.09000	119.5	.09500	120.4
.10960	110.0	.12960	143.6	.14960	156.0	.16960	175.4
.18960	180.2	.21960	189.4	.26460	205.0	.31460	212.5
.36460	216.4	.43960	237.2	.53960	234.7	.63960	234.2
.73960	229.1	.83960	221.1	.93960	213.6	1.03960	206.8
1.13960	200.8	1.23960	191.4	1.33960	189.9	1.43960	184.3
1.53960	179.0	1.63960	174.3	1.73960	170.2	1.83960	166.6
1.93960	163.4	1.99960	161.5				

EDS PRESSURE TIME HISTORY NUMBER 40							
40	46						
0.00000	0.0	.00500	0.0	.01000	0.6	.01500	3.6
.02000	8.6	.02500	16.4	.03000	28.7	.03500	44.2
.04000	62.0	.04500	81.8	.05000	97.0	.05500	110.7
.06000	118.3	.06500	116.5	.07000	113.7	.07500	114.3
.08000	116.2	.08500	119.7	.09000	122.9	.09500	121.5
.10960	122.3	.12960	152.3	.14960	164.4	.16960	180.7
.18960	186.2	.21960	155.2	.26460	209.3	.31460	217.2
.36460	220.9	.43960	244.6	.53960	243.3	.63960	240.3
.73960	234.7	.83960	226.3	.93960	218.6	1.03960	211.8
1.13960	205.9	1.23960	200.3	1.33960	194.7	1.43960	188.9
1.53960	183.5	1.63960	178.7	1.73960	174.5	1.83960	170.9
1.93960	167.6	1.99960	165.8				

EDS PRESSURE TIME HISTORY NUMBER 41							
41	46						
0.00000	0.0	.00500	0.4	.01000	4.9	.01500	17.3
.02000	35.4	.02500	62.3	.03000	99.9	.03500	141.6
.04000	178.8	.04500	212.5	.05000	237.4	.05500	255.2
.06000	266.2	.06500	271.3	.07000	272.1	.07500	271.7
.08000	271.9	.08500	269.2	.09000	268.2	.09500	270.8

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.04000	178.8	.04560	212.5	.07060	237.4	.05560	254.2
.06060	264.2	.06560	271.3	.07060	272.1	.07560	271.7
.08060	271.9	.08560	269.2	.09060	264.2	.09560	270.8
.10960	276.1	.12960	291.4	.14960	300.5	.16960	307.4
.16960	310.6	.21460	315.5	.26460	321.1	.31460	329.2
.36460	317.0	.43960	276.0	.53960	274.8	.63960	270.0
.73960	263.0	.83960	252.0	.93960	243.9	1.03960	256.9
1.13960	230.0	1.23960	224.5	1.33960	214.3	1.43960	211.9
1.53960	206.0	1.63960	200.8	1.73960	190.3	1.83960	192.4
1.93960	188.7	1.99960	186.8				

EOS PRESSURE TIME HISTORY NUMB-R 42

42	0.00000	0.0	7.7	.01000	35.7	.01500	93.8
.02000	175.0	.02500	271.6	.03000	382.2	.03500	491.1
.04000	570.3	.04560	629.1	.05060	651.2	.05560	663.6
.06060	669.2	.06560	673.2	.07060	676.8	.07560	679.7
.08060	682.0	.08560	685.9	.09060	689.1	.09560	691.3
.10960	693.4	.12960	699.3	.14960	704.2	.16960	705.0
.18960	713.1	.21460	706.8	.26460	701.2	.31460	681.7
.36460	672.1	.43960	654.3	.53960	619.1	.63960	608.7
.73960	583.9	.83960	549.4	.93960	524.8	1.03960	515.3
1.13960	504.1	1.23960	492.0	1.33960	479.0	1.43960	465.8
1.53960	457.3	1.63960	444.6	1.73960	436.2	1.83960	426.5
1.93960	420.8	1.99960	416.8				

NET THRUST  
ON CAVITY WALL  
(1/2 model) in  
direction of HOTGAS  
BREAK (lbs.)

APPROXIMATE NET ASYMMETRIC LOAD AND ASYMMETRIC LOAD + 90 DEGREES, FOR THIS LOAD CASE, BY TIME STEP

NSTEP#	1	TIME#	0.0000	ASYMMETRIC#	0.	ASYM+90DEG#	0.
NSTEP#	2	TIME#	.0050	ASYMMETRIC#	.90547E+05	ASYM+90DEG#	.28869E+05
NSTEP#	3	TIME#	.0100	ASYMMETRIC#	.39801E+06	ASYM+90DEG#	.15057E+06
NSTEP#	4	TIME#	.0150	ASYMMETRIC#	.95961E+06	ASYM+90DEG#	.41667E+06
NSTEP#	5	TIME#	.0200	ASYMMETRIC#	.17991E+07	ASYM+90DEG#	.87118E+06
NSTEP#	6	TIME#	.0250	ASYMMETRIC#	.26935E+07	ASYM+90DEG#	.15465E+07
NSTEP#	7	TIME#	.0300	ASYMMETRIC#	.41641E+07	ASYM+90DEG#	.24437E+07
NSTEP#	8	TIME#	.0350	ASYMMETRIC#	.54243E+07	ASYM+90DEG#	.35200E+07
NSTEP#	9	TIME#	.0400	ASYMMETRIC#	.62628E+07	ASYM+90DEG#	.46805E+07
NSTEP#	10	TIME#	.0456	ASYMMETRIC#	.66588E+07	ASYM+90DEG#	.60157E+07
NSTEP#	11	TIME#	.0506	ASYMMETRIC#	.66286E+07	ASYM+90DEG#	.71171E+07
NSTEP#	12	TIME#	.0556	ASYMMETRIC#	.63560E+07	ASYM+90DEG#	.80797E+07
NSTEP#	13	TIME#	.0606	ASYMMETRIC#	.57769E+07	ASYM+90DEG#	.89276E+07
NSTEP#	14	TIME#	.0656	ASYMMETRIC#	.56219E+07	ASYM+90DEG#	.97418E+07
NSTEP#	15	TIME#	.0706	ASYMMETRIC#	.53778E+07	ASYM+90DEG#	.10563E+08
NSTEP#	16	TIME#	.0756	ASYMMETRIC#	.52138E+07	ASYM+90DEG#	.11441E+08
NSTEP#	17	TIME#	.0806	ASYMMETRIC#	.50958E+07	ASYM+90DEG#	.12336E+08
NSTEP#	18	TIME#	.0856	ASYMMETRIC#	.49915E+07	ASYM+90DEG#	.13159E+08
NSTEP#	19	TIME#	.0906	ASYMMETRIC#	.49070E+07	ASYM+90DEG#	.13849E+08
NSTEP#	20	TIME#	.0956	ASYMMETRIC#	.48612E+07	ASYM+90DEG#	.14441E+08
NSTEP#	21	TIME#	.1006	ASYMMETRIC#	.48657E+07	ASYM+90DEG#	.16037E+08
NSTEP#	22	TIME#	.1296	ASYMMETRIC#	.49720E+07	ASYM+90DEG#	.17920E+08
NSTEP#	23	TIME#	.1496	ASYMMETRIC#	.47977E+07	ASYM+90DEG#	.19298E+08
NSTEP#	24	TIME#	.1696	ASYMMETRIC#	.42817E+07	ASYM+90DEG#	.20454E+08
NSTEP#	25	TIME#	.1896	ASYMMETRIC#	.43653E+07	ASYM+90DEG#	.21253E+08
NSTEP#	26	TIME#	.2146	ASYMMETRIC#	.45210E+07	ASYM+90DEG#	.22352E+08
NSTEP#	27	TIME#	.2646	ASYMMETRIC#	.40756E+07	ASYM+90DEG#	.23675E+08
NSTEP#	28	TIME#	.3146	ASYMMETRIC#	.37783E+07	ASYM+90DEG#	.24430E+08
NSTEP#	29	TIME#	.3646	ASYMMETRIC#	.37966E+07	ASYM+90DEG#	.24764E+08
NSTEP#	30	TIME#	.4396	ASYMMETRIC#	.31174E+07	ASYM+90DEG#	.25749E+08
NSTEP#	31	TIME#	.5396	ASYMMETRIC#	.28788E+07	ASYM+90DEG#	.25923E+08
NSTEP#	32	TIME#	.6396	ASYMMETRIC#	.25099E+07	ASYM+90DEG#	.25717E+08
NSTEP#	33	TIME#	.7396	ASYMMETRIC#	.23929E+07	ASYM+90DEG#	.25221E+08
NSTEP#	34	TIME#	.8396	ASYMMETRIC#	.22477E+07	ASYM+90DEG#	.24334E+08
NSTEP#	35	TIME#	.9396	ASYMMETRIC#	.21311E+07	ASYM+90DEG#	.23527E+08
NSTEP#	36	TIME#	1.0396	ASYMMETRIC#	.21125E+07	ASYM+90DEG#	.22745E+08
NSTEP#	37	TIME#	1.1396	ASYMMETRIC#	.20801E+07	ASYM+90DEG#	.22171E+08
NSTEP#	38	TIME#	1.2396	ASYMMETRIC#	.20292E+07	ASYM+90DEG#	.21549E+08
NSTEP#	39	TIME#	1.3396	ASYMMETRIC#	.19769E+07	ASYM+90DEG#	.21005E+08
NSTEP#	40	TIME#	1.4396	ASYMMETRIC#	.19230E+07	ASYM+90DEG#	.20398E+08
NSTEP#	41	TIME#	1.5396	ASYMMETRIC#	.18824E+07	ASYM+90DEG#	.19817E+08

SECTION 4: INTEGRATED  
NET THRUST DATA

NSTEP#	39	TIME#	1.3396	ASYMMETRIC#	.19769E+07	ASYM+90DEG#	.21005E+08
NSTEP#	40	TIME#	1.4396	ASYMMETRIC#	.19750E+07	ASYM+90DEG#	.20398E+08
NSTEP#	41	TIME#	1.5396	ASYMMETRIC#	.18870E+07	ASYM+90DEG#	.19817E+08
NSTEP#	42	TIME#	1.6396	ASYMMETRIC#	.18095E+07	ASYM+90DEG#	.19299E+08
NSTEP#	43	TIME#	1.7396	ASYMMETRIC#	.18217E+07	ASYM+90DEG#	.18853E+08
NSTEP#	44	TIME#	1.8396	ASYMMETRIC#	.17902E+07	ASYM+90DEG#	.18467E+08
NSTEP#	45	TIME#	1.9396	ASYMMETRIC#	.17623E+07	ASYM+90DEG#	.18115E+08
NSTEP#	46	TIME#	1.9996	ASYMMETRIC#	.17490E+07	ASYM+90DEG#	.17912E+08



```

WCZ WCC RC0PE 3.4.3 006G.000 08/07/79 8/115
15.30.48.00ENR047 FROM 700
15.30.48.18IP, INPUT, 00000012 IODBS, ***/NO, DC# 40.
15.30.48.55ENR047, DEN.
15.30.48.58MAP, 04, TPI, T100. * MANIPUL
15.30.49.00USEH(ENR004C,1)
15.30.50.00PROJECT, *110*001221.
15.30.51.00ACCN, F3275, *110*001221
15.30.52.00ROUTE, OUTPUT, DC#R, FC#IP, DEF.
15.30.52.00RTN, LRD, A.
15.30.57.00 701 CP SECONDS COMPILATION TIME
15.30.57.00MAP, OFF.
15.30.57.00LABEL, X, R, YSH#PA4224.
15.31.57.00MT24 VOLUME SERIAL NUMBER IS PA4224
15.31.57.00MT24 ASSIGNED TO X
15.32.00.00VSH# PA4224, NO ACCESS GRANTED
15.32.00.00LS READ HAS J#EXHWC#007
15.32.00.00REITION NUMBER 01
15.32.00.00RETENTION CYCLE 000
15.32.00.00CREATION DATE 79242
15.32.00.00REEL NUMBER 0001
15.32.00.00COPY#R, TAPE1, 5.
15.32.22.00UNLOAD, X.
15.32.22.00REWIND, TAPE1.
15.32.22.00LGO, PL=20000.
15.32.39.00END MAP
15.32.40.00ISOP, OUTPUT, 00000431 IODBS, ***/NO, DC# 40.
15.32.40.00ISEQ, ENTERED QUEUE 15.30.47 79327
15.32.40.00ISSM, 20.747 EXECUTION TIME
15.32.40.00ISS 60928 WORDS ( 215040 MAX USED)
15.32.40.00CPA 12.889 SEC.
15.32.40.00IU 07.857 SEC.
15.32.40.00CM 986.662 KMS.
15.32.40.00ISSN, 44.391 TOTAL SHUS NON-APPLICATION
15.32.40.00PP 41.631 SEC. DATE 11/23/79
15.32.40.00EJ END OF JOB, NO

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APPENDIX C  
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- <sup>2</sup> B&W 177-FA Owners Group — Effects of Asymmetric LOCA Loadings, Phase II Analysis, BAW-1621, Babcock & Wilcox, July 1980.
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- <sup>6</sup> Reactor Coolant System Hydrodynamic Loadings During Loss-of-Coolant Accident, BAW-10132P-A, Babcock & Wilcox, May 1979.
- <sup>7</sup> Reactor Coolant System Hydrodynamic Loadings During Loss-of-Coolant Accident, BAW-10132P-A, Supplement 1, Babcock & Wilcox, May 1979.
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- <sup>9</sup> A. Schwencer (NRC) to K. E. Suhrke (B&W), Letter, "Review of B&W Operating Experience of Reactor Internals Vent Valves," November 19, 1975.
- <sup>10</sup> Davis Bessie 1 Safety Analysis Report.
- <sup>11</sup> Rancho Seco Final Safety Analysis Report.
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- <sup>17</sup> Babcock & Wilcox Calculation 86-1107048-00, Enclosure With Letter 595-7086-12 to EDS Nuclear, February 19, 1980.
- <sup>18</sup> Regulatory Guide 1.60, Revision 1, U. S. Nuclear Regulatory Commission, December 1973.
- <sup>19</sup> J. M. Biggs, Introduction to Structural Dynamics, McGraw-Hill Book Co., New York (1974).
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- <sup>22</sup> J. W. Fisher and J. H. A. Struik, Guide to Design Criteria for Bolted and Riveted Joints, John Wiley & Sons, New York (1974).
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