

Pathfinder Decommissioning

Application for Type A Container — Reactor Vessel

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APPLICATION FOR APPROVAL OF PACKAGING FOR LSA MATERIAL IN TYPE A PACKAGING

DECOMMISSIONING OF THE PATHFINDER ATOMIC POWER PLANT

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for

NORTHERN STATES POWER COMPANY

October 1989

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Rev. 0

TABLE OF CONTENTS

Sect	100	20	ge
Exec	utive	Summary	ii
1.0	Gener	ral Information	- 1
	1.1	Introduction	-1
	1.2	Package Description.	-2
		1.2.1 Vessel and Internals.	-3
		1.2.2 Shielding, Impact Limiters and Tie-downs	-3
		1.2.3 Package Weight and Dimensions	-4
		1.2.4 Total Curie Content	-5
		1.2.5 Decay Heat Generation	-5
	1.3	Appendix	-6
		References	-6
2.0	Struc	tural Evaluation	-1
	2.1	Structural Design	-1
		2.1.1 Structural Description	
		2.1.2 Design Criteria	-3
	2.2	Weights and Center of Gravity	-6
	2.3	Mechanical Properties of Materials	-6
		2.3.1 Steel Materials	-6
		2.3.2 Impact Limiter	-7
		2.3.3 Brittle Fracture Evaluation	-7
	2.4	General Standards for All Packages	-8
		2.4.1 Minimum Package Size	-8
		2.4.2 Tamper-Proof Feature	-8
		2.4.3 Positive Closure	-8
		2.4.4 Chemical and Galvanic Reactions	-8
		2.4.5 Package Valves	-8
		2.4.6 Accessible Surface Temperature	-8
		2.4.7 Continuous Venting2	-8
		2.4.8 Lifting Devices	-8
		2.4.9 Tie-Down Devices2	-9
	2.5	Standards for Type B and Large Quantity Packaging2	-9
	2.0	Normal Conditions of Transport	-9
		2.0.1 heat	-9
		2.6.2 Columna External Draggura	10
		2.6.4 Increased External Dressure	10
		2.6.5 Vibration	10
		2.6.6 Water Spray	10
		2.6.7 Free Drop	11
		2.6.8 Corner Drop	13
		2.6.9 Compression	13
		2.6.10 Penetration2-	13
		2.6.11 Load Resistance 2-	14
	2.7	Hypothetical Accident Conditions2-	14
	2.8	Special Form	14
	2.9	Fuel Rods	14
	2.10	Appendix	15

Rev. 0

3.0	Thermal Evaluation
4.0	Containment
	4.2 Requirements for Normal Conditions of Transport4-2 4.2.1 Containment of Radioactive Material
	4.3 Containment Requirements - Hypothetical Accident Conditions4-4
5.0	Shielding Evaluation
	5.2 Source Specification
	5.3 Model Specification
	5.4 Shielding Evaluation
6.0	Criticality Evaluation
7.0	Operating Procedures
8.0	Acceptance Tests and Maintenance Program
	8.2 Maintenance Program

EXECUTIVE SUMMARY

The Fathfinder Atomic Power Plant is being decommissioned. The reactor vessel and internals will be removed in one piece and shipped for disposal at Richland, WA, using the reactor vessel as the shipping container. The reactor vessel package includes two inches of steel circumferential shielding around the core region to maintain exposures with NRC and DOT shipping requirements. The package uses a honeycomb structure external impact limiter to meet the Normal Conditions of Transport free drop analyses. The package will be shipped by rail from Sioux Falls, SD to Richland, WA.

This application by Northern States Power Company (NSP) to the Nuclear Regulatory Commission (NRC) contains the necessary information identified in NRC Regulatory Guide 7.9, "Standard Format and Content of Part 71 Applications for Approval of Packaging of Type B, Large Quantity, and Fissile Radioactive Material" (Ref. 1.1) to gualify the vessel as a Type A container for Low Specific Activity (LSA) material. The application contains no exceptions to the NRC regulations identified in Title 10 CFR Part 71, "Packaging and Transportation of Radioactive Materials", (Ref. 1.2).

1.0 GENERAL INFORMATION

1.1 Introduction

Northern States Power Company (NSP) plans to decommission the Pathfinder Atomic Power Plant's (Pathfinder) fuel handling building (FHB) and reactor building (RB). Pathfinder is located in South Dakota, approximately 5.5 miles northeast of the city of Sioux Falls. It is owned and was operated as a nuclear facility by NSP of Minneapolis, Minnesota. The nuclear steam supply system (NSSS) was a 66 MWe boiling water type nuclear reactor designed by Allis-Chalmers Manufacturing Company of Milwaukee, Wisconsin.

Pathfinder was operated as a nuclear facility by NSP from 1964 through 1967. Starting in 1968, the plant was repowered with fossil fueled boilers and the nuclear portions of the plant were partially decommissioned and placed in a safe storage (SAFSTOR) configuration. Decommissioning was accomplished under an amendment to the original nuclear plant operating license, DPR-11. Following completion of the partial decommissioning program in 1971, the 10 CFR Part 50 license was surrendered upon amendment of a by-product material possession license under the provisions of 10 CFR Part 30.

During the SAFSTOR decommissioning program, the NSSS was dismantled to the extent necessary to render it inoperable and incapable of being restored to service. The reactor vessel was drained and connected to a vacuum pump to remove all residual water from nondrainable cavities. The control rods and blades were placed in the vessel for disposal, and the reactor vessel was filled with gravel. The internals were left in place. The vessel head was reinstalled and bolted to the vessel flange using the vessel head bolts.

Two vessel removal alternatives were considered; segmentation of the vessel and internals and transport for disposal in shielded shipping containers, and one-piece removal and disposal. The segmentation alternative would require extensive remote tooling and individual segment handling to cut and load the shipping casks for disposal. The cutting process would generate large quantities of smoke and debris which would require extensive contamination control measures such as control access tents, high efficiency particulate air (HEPA) filters and remote handling devices. There would be additional exposure to workers owing to the extensive time required to segment, load and ship each piece. The selected alternative of one-piece shipment does not require extensive tooling, handling contamination controls or high worker radiation exposure. The vessel and internals package contains 476 curies of radioactivity as activated and contaminated material. This number of curies does not warrant the effort to segment and package the vessel and internals. Based on

the vessel activation and shielding analysis, the vessel and internals qualify as Low Specific Activity material and the package qualifies as a Type A container.

To prepare the vessel for shipment, the vessel will be filled with grout to fix the gravel in place, lifted from the reactor cavity, moved through the open reactor building dome and laid on its side within a temporary enclosure. The external asbestos insulation will be removed and the vessel certified clean of asbestos. A cylindrical steel shield will be welded to the vessel shell. An impact limiter consisting of Hexcel energy absorbing material will be installed on the vessel shell surface to protect the vessel during normal condition of transport. The vessel package will be lifted horizontally and placed on saddles installed on a flatbed railcar for transport to the US Ecology, Inc. burial facility near Richland, WA.

When fully prepared the vessel package will weigh 291 tons not including tie-downs or other supports. The package will be transported by rail from the Pathfinder site to Richland, WA. The Type A container package will be shipped in an exclusive use vehicle (railcar), but does not require a special carrier; therefore the package will be transported as normal mixed freight. The reactor vessel package will be unloaded from the railcar and transported a short distance to the burial site.

This shipping package application report evaluates the adequacy of the Pathfinder vessel package to meet the applicable requirements of the NRC contained in Title 10 CFR Fart 71 (Ref. 1.2). Authorization is requested to approve this package for a one-time, single trip shipment between the Pathfinder site in Sioux Falls, SD and Richland, WA.

1.2 Package Description

The Pathfinder package is shown on Figure No. 1.1, and is based on TLG Engineering, Inc. Drawing No. N04-22B-001, Pathfinder Vessel Transport Package. The package is essentially a cylindrical container and will consist of the reactor vessel, internals, internal grout, steel shielding and impact limiter. The internal steel components of the package are neutron activated and include a variety of radionuclides, primarily cobalt-60 (Co-60), iron-55 (Fe-55), nickel-63 (Ni-63) and nickel-59 (Ni-59). A small amount of surface contamination exists on the interior surfaces of the vessel, but will be generally bound to the surfaces by the grout.

The total radioactivity is 476 Curies. However, the majority of the package source term consists of activated base metal that is not readily dispersible. The remaining source term is estimated to be 95 millicuries consisting of a thin corrosion film that is bound to the surfaces of the Pathfinder vessel and internals and is further

bound by the grout. The maximum specific activity of the package is approximately 0.248 millicuries per gram. Since this value is less than 0.3 millicuries per gram the package qualifies as a Low Specific Activity (LSA) package. In accordance with 10 CFR 71.52, the package is exempt from the additional requirements for Type B packages, and will be qualified as a Type A package.

1.2.1 Vessel and Internals

The major components of the reactor vessel and internals are shown in Figure 1.2. The reactor vessel is fabricated from 3-inch thick carbon steel ASME A212 Grade B plate with integrally bonded (Lukens clad) 304L stainless steel cladding. The flanges and nozzle forgings conform to the ASTM A105 Grade II and are weld overlaid with stainless steel having a chemistry similar to Type 304 except that maximum free carbon content at the clad surface is below 0.050 per cent. The overlay is stabilized with a small amount of niobium.

The vessel head is fabricated from the same carbon and stainless steel cladding materials as the vessel shell. The head is secured to the vessel flange by the original 48 head studs and nuts.

The vessel internals are all fabricated from Type 304L stainless steel with the exception of the boiler boxes which are Zircaloy-2. The other principal contents of the vessel are the control rods and blades that were disposed of in the reactor vessel. These components were fabricated from boron stainless steel and contain boron pellets.

At the recommendation of the Atomic Energy Commission at the time of the initial Pathfinder decommissioning, the vessel was filled with one-quarter inch diameter pea gravel. The pea gravel apparently filled all cavities uniformly with the exception of one area at the side of the vessel. At this location there is a "hot spot" with a radiation reading of about 600 mR/hour, whereas all other areas at the same elevation are about 150 mR/hour. To stabilize the gravel and to fill the apparent void spot, the vessel will be pumped full with a grout. Other than the filling of the apparent void, no direct credit was taken for the shielding effectiveness of the grout in the shielding analysis.

1.2.2 Shielding, Impact Limiter and Tie-downs

Based on the shielding analysis of the vessel package with a 25 per cent void fraction in the gravel (no credit for the grout), two inches of steel shielding will be added to the vessel package to meet the NRC and DOT transport regulations. The shielding will be made up from a 1-3/4 inch thick shell plate welded to the vessel shell outside diameter, extending three feet below the elevation of the core bottom and five feet above the elevation of the core top. Figure 1.1 shows the shield extending six feet below and eight feet above the core midplane. The remaining 1/4 inch of steel shielding will be fabricated in a cylindrical shape to form the outer shell for the Hexcel impact limiter. The two inches of steel shielding is sufficient to reduce the external exposure rate to less than 10 mRem/hour at two meters (without credit for the grout), assuming a 25 percent void fraction in the gravel.

The impact limiter is used to absorb the energy of impact from the postulated one foot drop during normal conditions of transport. The impact limiter will be fabricated from pre-crushed Hexcel (TM of the Hexcel Corporation, Dublin, CA); a honeycomb configuration of aluminum bonded to the vessel surface. Two types of Hexcel will be used; a low density Hexcel configuration bonded to the vessel shell region extending over the entire length of the vessel. In the region where the 12 inch wide vessel support saddles will be located (at the vessel upper and lower shell spring lines), a higher density Hexcel will be used to compensate for the loss of low density Hexcel in that area. The Hexcel will be bonded to the vessel, and then covered by a one-quarter inch thick steel cylindrical plate to protect the Hexcel from inadvertent damage. The minimum Hexcel thickness is 8 inches located at the center region of the shell, surrounding the 1-3/4 in thick shielding section. The Hexcel thickens to approximately 11 inches for the remaining region of the vessel shell.

The tim-downs used for the package shipment will consist of two hoop straps fabricated from wire rope or steel bands. The hoop straps will be attached to the support saddles or to the railcar floor structural member supports to ensure there will be adequate support during transport. The vessel will rest on the two circumferential saddles located at the vessel shell spring lines. There will be no direct attachments to the vessel shell or Hexcel material for vessel tie-down except for the circumferential weld of the shield to the vessel.

1.2.3 Package Weight and Dimensions

The overall package weight of the vessel, internals, gravel, grout, shield, and impact limiter is 582,000 lbs, or 291 tons. Since the saddles and tie-downs are not attached to the package directly, no additional weight is included herein. The weights of the individual major components are shown in Table 1.1.

The overall package dimensions are shown on Figure 1.1, and are based on TLG Engineering, Inc. Drawing No. N04-22B-001. The overall length is 35ft-4 1/2in, and the overall diameter is 13ft-2in. The other principal dimensions are shown in Figure 1.1.

1.2.4 Total Curie Content

The Pathfinder package contains both activated and contaminated materials. The reactor vessel and internals activated inventory constitute most of the radioactivity of the package. The total activation inventory is 476 curies, consisting primarily of Co-60, Fe-55, Ni-59 and Ni-63. In addition, approximately 95 millicuries of the same radionuclides are present in the form of surface contamination on the interior surfaces of the internals. Table 1.2 lists the major components of the vessel and internals and their activity (in Curies) by radionuclide.

The principal radionuclide of interior surface contamination is Co-60 based on data taken of scraping samples from the feedwater system piping attached to the vessel. This sample point was judged to be a conservative representation of the vessel source of surface contamination. The individual quantities of surface contamination are not shown on Table 1.2 because the amounts are so small relative to the activation inventory.

1.2.5 Decay Heat Generation

The total amount of decay heat generation for the vessel and internals package was calculated from the activation analysis to be 4.67 watts. This amount of decay heat is insignificant, and need not be considered further in this analysis.



APPENDIX 1.3



- 1.1 NRC Regulatory Guide 7.9, "Standard Format and Content of Part 71 Applications for Approval of Packaging of Type B, Large Quantity, and Fissile Radioactive Material", Revision 1, January, 1980.
- 1.2 Title 10 Code of Federal Regulations, Part 71, "Packaging and Transportation of Radioactive Materials", January 1, 1989.





1-7

Rev. 0

Rev. 0



Figure 1.2 Pathfinder Vessel and Internals

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TABLE 1.1 PATHFINDER PACKAGE CALCULATED WEIGHTS

Component	Calculated Weight, 1bs
Reactor Vessel	155,000
Vessel Internals	43,000
Shielding/Impact Limiter	64,000
Gravel and Grout	320,000
GROSS WEIGHT	582,000 lbs 291 tons

TABLE 1.2 ESTIMATED INVENTORY OF RADIONUCLIDES ON JANUARY 1, 1990

Component	H3	C14	Fe55	Co60	N159	N163	Nb94	Tc99	Eu152	Eu154	Others	Total Curies
Superheater baffle	0.08	0.03	4.58	34.40	0.19	21.97			0.02			61.3
Superheater fuel insul. tubes	0.15	0.06	8.77	65.92	0.35	42.11			0.05			117.4
Superheater support plate												<0.01
Superheater control rods			0.04	0.32		0.21						0.6
Boiler fuel boxes	0.01	0.03	0.84	7.19	0.04	4.35					0.86	13.3
Boiler shroud	0.15	0.06	9.59	51.61	0.39	46.84					0.01	108.6
Boiler hold down structure	0.01		0.61	4.58	0.02	2.93						8.1
Boiler CR tubes/remain. struct.	•••		0.49	3.69	0.02	2.36	•••				0.01	6.6
Builer element poison shims	0.03	0.01	1.88	14.14	0.08	9.03			0.01			25.2
Boiler control blades	0.11	0.04	6.39	48.03	0.26	30.68			0.03			85.5
Instrumentation/sample holders			0.48	3.58	0.02	2.29					0.01	6.4
Boiler grid plate	0.04	0.02	: 52	18.96	0.10	12.11			0.01			33.8
Steam separators & supports			0.54	2.93	0.02	2.66					0.01	6.2
Feedwater ring & supports			0.15	0.83		0.75				***	0.01	1.7
Vessel cladding			0.06	0.34		0.31						0.7
Vessel	•••	••••	0.41	0.14		0.09						0.6
	•••••		•••••									
Total by isotope (curies)	0.59	0.25	37.36	256.66	1.49	178.68	<0.01	<0.01	0.13	<0.01	0.93	476.1
Percent of total by isotope	0.12%	0.05%	7.85%	53.91%	0.31%	37.53%	<0.01%	<0.01%	0.03%	<0.01%	0.19%	

2.0 STRUCTURAL EVALUATION

2.1 Structural Design

The purpose of this chapter is to describe and evaluate the structural aspects of the Pathfinder package for compliance with the requirements specified in 10 CFR 71 (Ref. 1.2). The principal structural engineering design features are identified, and the packaging components and systems important to safety are described and evaluated. Analyses results are presented which demonstrate that adequate protection measures are being taken to assure safe transport of the Pathfinder package for Normal Conditions of Transport.

The basic acceptance criteria for this package are that it must meet the external radiation requirements of the NRC as defined in 10 CFR 71 paragraph 71.47 (a) through (d). The structural criteria for this package are that it must meet the requirements of the NRC as defined in 10 CFR Part 71.71, paragraphs (a) through (c). The package must be shown to maintain integrity under the Normal Conditions of Transport, and not breach containment and lose its contents under these postulated test conditions. The radioactivity contained in the Pathfinder package is in the form of activated materials and internal surface contamination, further bound to the surface by grout pumped inside the vessel under pressure to fill voids. The package contains no contaminated liquids, explosive mixtures or potential aerosol particulates that could be considered radiological hazards. The analyses presented in this chapter show that the Pathfinder package can be safely transported from Sioux Falls, SD to Richland, WA without loss of integrity of the container.

The containment boundary consists of the thick reactor vessel shell, reactor closure head and nozzle closure plates. An outer 1/4 inch thick shell is included to contain the Hexcel.

Based on the finite element analysis performed on the Pathfinder package with the Hexcel impact limiter, the vessel shell will not deform plastically during the one foot free drop case. The elastic deformation that occurs is well within the allowable stress limits of the vessel materials. There will be no loss of the external shielding during the one foot drop, although the outer Hexcel 1/4 inch thick shell will deform plastically. Nevertheless, this outer shell will retain its shielding capability and protect the public from exposure.

The primary method of energy dissipation for the drop events is the crushing of the pre-crushed Hexcel impact limiter. Analyses show that the Hexcel will absorb the drop impact and protect the vessel shell from damage. The analyses consisted of finite element ANSYS (Ref. 2.1) computer runs to determine the maximum stresses in the vessel and hand calculations to verify the results.

The analytically determined margins of safety for each of the transport conditions are summarized in Table 2.1 for the Normal Conditions of Transport.

2.1.1 Structural Description

The principal structural elements of the Pathfinder package are the reactor vessel, vessel head and the Hexcel impact limiter. The steel shield surrounding the vessel does not have a structural role in the package design. The circumferential weld of the shield to the vessel is used as part of the package tie-down.

Vessel, Nozzles and Lower Head

The 3-inch thick reactor vessel has a lower hemispherical head to which is welded a lower cylinder assembly. The hemispherical head contains three pump suction nozzles and one feedwater nozzle. The lower cylinder assembly contains three pump discharge nozzles and one steam outlet nozzle extending through the bottom of the cylinder assembly. There is a liquid level nozzle and an instrument nozzle connected to the shell of the vessel. All nozzles will be cut close to the vessel shell, and nozzle cover plates will be welded over the openings. In addition, there are five pad assemblies welded to the vessel shell which support the vessel vertically within the vessel cavity and four plates for lateral support. These pads and plates will be cut close to the vessel prior to the installation of the steel shielding. The reactor vessel is fabricated from ASTM A212 Grade B plate with integrally bonded 304L stainless steel cladding 1/4-inch thick. The flanges and nozzle forgings conform to the ASTM A105 Grade II specification and are weld overlayed with stainless steel having a chemistry similar to Type 304 except that maximum free carbon content at the clad surface is below 0.050 per cent. The overlay is stabilized with a small amount of niobium. The original design pressure and temperature for the vessel were 700 psi and 500 deg F, respectively. The vessel and internal components were designed and constructed in accordance with the ASME Boiler and Pressure Vessel Code Section VIII and applicable special nuclear code cases.

Closure Head

The closure head is fabricated from the same material as the vessel shell and lower head. The head is secured to the reactor vessel with 48 closure studs fabricated from ASTM A437 Grade B4B, and the nuts fabricated from ASTM A437 Grade C4C. There are 20 control rod drive nozzles in the upper closure head, four lifting pipes used for head removal, one liquid level nozzle and one instrumentation nozzle. Each of these nozzles will be capped and seal welded closed prior to lifting the vessel from the reactor cavity.

Rev. 0

Gravel and Grout

As noted earlier, the vessel was filled with 1/4 inch diameter pea gravel during the decommissioning activities associated with placing the facility in safe storage. Approximately 66 cubic yards of gravel, weighing 214,000 lbs (average density of 120 lbs/cu ft) were placed in the vessel. The gravel therefore has a void fraction of 23.4 percent based a theoretical solid density of 156.9 lbs/cu ft, or a void volume of 15.44 cu yds (.234 X 66 cu yds = 15.44 cu yds).

The estimated empty volume of the vessel less the internals is approximately 75.6 cubic yards. The gravel occupies only 66 cu yds of the 75.6 cu yds, leaving 9.6 cu yds empty. Approximately 25.04 cubic yards of grout needs to be added to the vessel to fill all available voids in the vessel and the gravel. This grout will be pumped in under pressure.

Impact Limiter

The impact limiter is constructed from Hexcel. Hexcel is a honeycomb energy absorbing material to absorb the impact of the one foot drop of the vessel. The honeycomb material will be fabricated from aluminum, and will be pre-crushed to eliminate any initial impact force peaks being transmitted to the vessel shell. The Hexcel will be bonded to the vessel shell and a 1/4 inch thick plate shell attached on the outside of the package to contain the Hexcel. All closure seams of the 1/4 inch thick shell will be welded closed. A minimum of 8 inches of Hexcel will be used around the cylindrical shell of the vessel. In the region where the two vessel saddle cutouts are located, high density Hexcel will be used to compensate for the loss of energy absorbing capacity at the saddles. The Hexcel material selected for this area will be 1/4-5052-0.004 with a crush strength of 725 psi (Ref. 2.2). In the remainder of the impact limiter, 3/8-5052-0.0025 Hexcel will be used with a crush strength of 180 psi.

The Hexcel thickness will vary to match the outside diameter of the vessel and steel shield so as to create a smooth package exterior. The configuration of the Hexcel arrangement is shown in Figure 1.1.

2.1.2 Design Criteria

Regulatory Requirements

The design criteria for the Pathfinder package are specified in NRC regulations Title 10 CFR Part 71. The activation inventory in the package is 476 curies. No individual vessel or internals component exceeds 0.3 millicuries per gram of material. Therefore, in accordance with paragraph 71.4 the package qualifies as Low Specific Activity (LSA) waste. The quantities of individual radionuclides exceed the limits for Type A packages specified in Table A-1 of Part

71 for special form and normal form values of A1 and A2, respectively. However, in accordance with paragraph 71.52 a package need not satisfy the requirements of paragraph 71.51 (Additional requirements for Type B packages) if it contains only Low Specific Activity material and is transported as exclusive use, but is subject to paragraph 71.41 through 71.47 of this part, including paragraph 71.43(f). Therefore, the package will be designed as a Type A package with contents that meet LSA quantity materials. Paragraph 71.43(f) requires the package be designed, constructed and prepared for shipment so that under the tests specified in paragraph 71.71 (Normal Conditions of Transport) there would be no loss or dispersal of radioactive contents, no significant increase in external radiation levels, and no substantial reduction in the effectiveness of the packaging.

Type A Package Requirements

To meet the requirements for a Type A package, the Pathfinder package must be evaluated in accordance with the Normal Conditions of Transport identified in paragraph 71.71 (a) through (c). Paragraph (a) identifies the test specimens and sequence for the tests specified in this section. Paragraph (b) identifies the initial conditions for all tests, namely that the ambient temperature must be between -29 deg C (-20 deg F) and +38 deg C (100 deg F) which is most unfavorable for the feature under consideration. The initial internal pressure must be considered to be the maximum normal operating pressure, unless a lower internal pressure consistent with the ambient temperature considered to precede and follow the tests is more unfavorable. Paragraph (c) provides the conditions and tests. These conditions and tests are as follows:

(c) (1) Heat An ambient temperature of 38 deg C (100 deg F) in still air, and insolation of 400 g cal/cm² for a 12-hour period (curved surfaces).

(c)(2) Colá An ambient temperature of -40 deg C (-40 deg F) in still air and shade.

(c)(3) Reduced External Pressure An external pressure of 24.5 kilopascal (3.5 psi) absolute.

(c)(4) Increased External Pressure An external pressure of 140 kilopascal (20 psi) absolute.

(c) (5) Vibration Vibration normally incident to transport.

(c)(6) Water Spray A water spray that simulates exposure to rainfall of approximately five cm (two in.) per hour for at least one hour. (c)(7) Free Drop

A free drop through a distance of 0.3 meters (1 foot) onto a flat, essentially unyielding, horizontal surface, striking the surface in a position for which maximum damage is expected.

(c)(8) Corner Drop
This test does not apply.

(c)(9) Compression This test does not apply.

(c)(10) Penetration

Impact of the hemispherical end of a vertical steel cylinder of 3.2 cm (1 1/4 in.) diameter and six kg (13 1b) mass, dropped from a height of one m (40 in.) onto the exposed surface of the package which is expected to be most vulnerable to puncture. The long axis of the cylinder must be perpendicular to the package surface.

ASME Code

The ASME Boiler and Pressure Vessel Code (ASME), Section III (Ref. 2.3) design requirements will be used to evaluate the Pathfinder package integrity for internal or external pressure and temperature effects during Normal Conditions of Transport.



Welding Research Council Bulletin WRC 107

The Welding Research Council Bulletin (WRC) 107 (Ref. 2.4) will be used to verify the Pathfinder package analysis performed by finite element methods.

Brittle Fracture

Brittle fracture analysis will be evaluated in accordance with "Recommendations for Protecting Against Failure by Brittle Fracture in Ferritic Steel Shipping Containers up to Four Inches Thick", NUREG/CR-1815 (Ref. 2.5).

Penetration Analysis

The penetration analysis will be performed in accordance with the Ballistic Research Lab (BRL) and Stanford Research Lab (SRL) equations. These equations are described in Bechtel Topical Report 9A, Revision 2, "Topical Report - Design of Structures for Missile Impact", Bechtel Power Corporation, San Francisco, CA, September, 1974 (Ref. 2.6).

Impact Limiter Requirements

The impact limiter will be designed in accordance with Hexcel Design Bulletin TSB122 (Ref. 2.7). No more than 70 per cent of the Hexcel honeycomb thickness is available for crushing. This provides a factor of safety to ensure the honeycomb material does not bottom out and transfer the impact load to the vessel shell. The total depth of crushing should not approach any nozzle or vessel attachment projections from the vessel surface.

2.2 Weights and Center of Gravity

The total weight of the vessel, internals, grout, shield and impact limiter is 582,000 lbs, or 291 tons. The distribution of the weights calculated in one foot increments along the package length, and center of gravity calculations are shown in Appendix 2.10.1. The total weights of the individual components is shown in Table 1.1. The center of gravity (CG) of the package has been determined to be located on the longitudinal centerline, approximately 16ft-4in. from the outside face of the package at the reactor vessel head.

2.3 Mechanical Properties of Materials

The primary structural materials of the Pathfinder package are the carbon steel vessel and the Hexcel aluminum honeycomb. The gravel and grout materials are not relied upon for their structural characteristics and will not be described here. The mechanical properties of the materials used for the analyses of the Pathfinder package are discussed in the following sections.

2.3.1 Steel Materials

Various types and grades of steels were used for fabrication of components of the Pathfinder vessel and heads. A tabulation of the specific types of steel materials used, their application and their material properties is shown in Tables 2.2 and 2.3. These tables were reproduced from the Allis-Chalmers Report No. ACNP-62025 (Ref. 2.8) which describes the reactor vessel materials, fabrication and inspection.

Figure 2.1 shows the Allis-Chalmers Manufacturers' Data Report For Unfired Pressure Vessels, Form U-1A. This form identifies the materials and thicknesses of the vessel and head materials and provides the certificate of shop inspection.

Table 2.4 shows the specified minimum mechanical properties and design stress intensity values, respectively, of steel materials for ASME and ASTM materials specifications. Table 2.5 shows the actual design stress intensity values for the vessel and associated component steels for the Pathfinder vessel. Because there was only 83 effective full power days (EFPD) the shift of the nil deductibility transition (NDT) temperature is negligible.

2.3.2 Impact Limiter

The impact force deformation curve is based on the Hexcel information bulletins for the various products available. The crushing strength and other structural characteristics for Hexcel type 1/4-5052-0.0015 and 3/8-5052-0.0025 (Ref. 2.2) is given in Table 2.6. A typical force-deformation curve for Hexcel is shown in Figure 2.2.

2.3.3 Brittle Fracture Evaluation

The requirements for brittle fracture evaluation are identified in NUREG/CR-1815, UCRL-53013, RT, "Recommendations for Protecting Against Failure by Brittle Fracture in Ferritic Steel Shipping Containers Up to Four Inches Thick", (Ref. 2.5). Table 6 of NUREG/CR-1815 shows the fracture toughness requirements for Category III required degree of safety. Category III provides for "Adequate margin of safety" with the required amount of fracture toughness "Sufficient to prevent fracture initiation at minor defects typical of good fabrication practices". Since the Pathfinder vessel was designed and fabricated for high pressure power reactor application, it is reasonable to assume it was fabricated in accordance with "good fabrication practices".

The NUREG/CR-1815 criteria for meeting toughness requirements is identified in Table 6, therein. Table 6 criteria for meeting toughness requirements are that if the vessel steel up to 4 inches thick was tested to show the Charpy impact test data Cv was greater than 15 ft-lb at 10 deg F test temperature, the steel has sufficient fracture toughness and adequate margin of safety. Based on the ACNP Report 62025 (Ref. 2.8) Table II, therein, as reproduced in Table 2.3, herein, the Charpy impact test for all components of the vessel and internals exceeded 15 ft-lb at 10 deg F. Therefore, the Pathfinder package has sufficient fracture toughness to provide adequate margin of safety. The results of the brittle fracture evaluation are shown in Appendix 2.10.6-A.

2.4 General Standards for All Packages

The general package standards in 10 CFR 71, Subpart E, Paragraph 71.43 (a) through (h), have been complied with as demonstrated in the following paragraphs.

2.4.1 Minimum Package Size

The Pathfinder package meets the package size criterion of minimum overall dimension not less than 4 in.

2.4.2 Tamper-Proof Feature

The containment system of the package consists of the welded nozzle closure plates and the 48 head closure studs and nuts. In addition the 1/4 inch Hexcel outer enclosure is welded at all joints and as such, no tamper-proof features are required.

2.4.3 Positive Closure

The outer boundary of the package (the 1/4 inch Hexcel enclosure) is welded at all joints so the system can not be inadvertently opened.

2.4.4 Chemical and Galvanic Reactions

The materials used in the package will not cause any significant chemical galvanic or other reactions. Since this is a one-time shipment, the long-term interactions of the materials are not of concern.

2.4.5 Package Valves

The package does not include any valves. There is no pathway for radioactive material to escape.

2.4.6 Accessible Surface Temperature

The package has a negligible internal heat source of 4.67 watts. Accordingly, accessible surface temperatures are not expected to be greater than the railcar surfaces on which the package is loaded.

2.4.7 Continuous Venting

There are no vents in the package.

2.4.8 Lifting Devices

The reactor vessel will be lifted from the biological shield cavity using a temporary lifting rig. Six of the existing head studs and nuts will be removed and replaced with longer studs. A temporary lifting fixture will be attached with the new studs and nuts, the



vessel lifted from the cavity and placed on its side outside the reactor building to prepare it for shipment. Prior to placement on the railcar, the temporary lifting rig will be removed and the vessel head studs retorqued and welded for a strong closure. All subsequent lifts will be made by lifting slings with the package in the horizontal position.

2.4.9 Tie-Down Devices

Tie-downs are not a part of the Pathfinder structural package. The package will be held in saddles with tie downs. Axial loads will be reacted by a key welded to the shield plate which will fit in a keyway in the saddles. The shield plate will be welded to the vessel shell using a continuous 11/16-inch circumferential fillet weld. This weld will be designed for a 10 g axial load.

2.5 Standards for Type B and Large Quantity Packaging

Not applicable.

2.6 Normal Conditions of Transport

This section demonstrates that the Pathfinder package is structurally adequate for the Normal Conditions of Transport as defined in Section 71.71 of 10 CFR 71. The 10 CFR 71 definition of structural adequacy is addressed in paragraph 71.51 (a)(1) which states that, "there would be no loss or dispersal of radioactive contents, as demonstrated to a sensitivity of 10⁻⁶ A2 per Hour, no significant increase in external radiation levels, and no substantial reduction in the effectiveness of the packaging". To assure that this qualitative definition is met, the criteria, as discussed in Section 2.1.2, have been applied.

2.6.1 Heat

For Normal Conditions of Transport, hot environment, 10 CFR 71 specifies exposure to an ambient temperature of 100 deg F in still air, with insolation of 400 g cal/cm² for a 12 hour period (for curved surfaces). Since there is negligible internal heat source, and there are no constraints to the vessel expansion and contraction the vessel stresses are assumed to be low. No calculations are provided.

2.6.2 Cold

For Normal Conditions of Transport, cold environment, 10 CFR 71 specifies an ambient temperature of -40 deg F in still air and shade. Since there is negligible internal heat source, and there are no constraints to the vessel expansion and contraction, the temperature gradient is small and the stresses are assumed to be lo^o There is no liquid that could freeze.

A worst case scenario is evaluated in Section 3.0 for an instantaneous reduction in temperature from 100 deg F to -40 deg F. The vessel stresses are within the design stress intensity limits for the vessel material.

2.6.3 Reduced External Pressure

The bounding case for this normal condition of transport is specified in 10 CFR 71.71 (c)(3), which indicates a 3.5 psi absolute pressure. A finite element analysis of the reactor vessel as an axisymmetric shell model indicated maximum stress intensity of 248.0 psi, as compared to a vessel material allowable stress intensity of 17,875 psi. Therefore, the margin of safety (based on the finite element analysis and ASME Code calculations) is (17,875/248) -1 = 71.08. The detailed ASME Code calculations and finite element analysis are shown in Appendix 2.10.2.

2.6.4 Increased External Pressure

The 10 CFR 71 Subpart F, paragraph 71.71 (c)(4) requires evaluation for an external pressure equal to 20 psi absolute. Using the ASME Code, Section III, paragraph NB-3133 (Ref. 2.10), the minimum allowable external pressure is 437.0 psi versus the required external pressure of 20.0 psi. Therefore, the margin of safety is 437/20 - 1= 20.9. The ASME Code evaluation for external pressure is shown in Appendix 2.10.3.

2.6.5 Vibration

The design criterion for rail transport vibration load was based on discussions with Burlington Northern Railroad wherein the railcar structure is designed for maximum buff and draft loads of 500,000 lbs and impact loads of 1,250,000 lbs. These criteria are based on the Association of American Railroad requirements. Accordingly, the g force would be 2.2 longitudinal.

10 CFR 71 paragraph 71.45(b)(1) requires tie-down devices attached to the package to be designed for 2g's vertical, 5g's transverse and 10g's longitudinal.

The only tie-down attachment to the package is the circumferential weld between the shield plate and the vessel shell. This weld will be designed for a log axial load.

2.6.6 Water Spray

The water spray condition will not have a significant effect on the Pathfinder package.

2.6.7 Free Drop

The Pathfinder package weighs 291 tons. Subsection F of 10 CFR 71 requires that a package of this mass be analyzed for a drop of one foot onto a flat, essentially unyielding, horizontal surface, striking the surface in a position for which maximum damage is expected. During transport the package will always be oriented in the horizontal position; i.e., the centerline axis of the package will be horizontal. Because of the unique size, weight, mass and bulk of the package, no other orientation is considered feasible during normal transport conditions. Consequently, the free drop condition is a one-foot side drop of the package on an essentially unyielding horizontal surface.

Two cases were evaluated: (1) drop of the package along its length, and (2) impact of the package at its edge, wherein 50 per cent of the package weight and kinetic energy is transmitted to one edge of the package. Drop of the package at its lower edge was considered to be more critical as it is heavier due to the core internals weight. For Case 1, the effective weight is taken as that of the heaviest one-foot wide section of the vessel and its internals along its length. It should be noted that for the side drop (Case 1), the weight of the protruding parts of the package (the upper and lower vessel heads) will be transmitted to the impact limiter at the edge of the vessel. The impact limiter end sections are analyzed for a significantly larger weight and kinetic energy in Case 2. Therefore, the maximum tributory (effective) weight for Drop Case 1 will be that of the heaviest one-foot wide section of the vessel and internals along its length. Figure 2.3 shows the orientation of these two drop cases. Maximum tributary weight, velocity and kinetic energy of impact associated with the two cases are given in Table 2.7. Refer to Appendix 2.10.4 for basis of the calculations.

2.6.7.1 Analytical Methods

This section describes the analytical methodology used to demonstrate the structural adequacy of the package during a side or edge drop event. The analytical method is based on a step-by-step evaluation of external energy, resistance offered and energy absorbed by the impact limiter. The maximum stresses in the reactor vessel wall are evaluated at each incremental depth of crushing. The Section 2.6.7.2 describes the calculation of deformation and forces generated during the drop event. Section 2.6.7.3 describes the ANSYS three dimensional finite element models used for the detailed evaluation of maximum stresses in the package.

2.6.7.2 Impact Limiter Resistance/Deformation Behavior

As described in Section 1.2, the package includes a circular energy absorbing impact limiter surrounding the vessel. This impact limiter is designed to protect the package during a one-foot drop event. The impact limiter consists of pre-crushed Hexcel energy absorbing material. For purposes of analysis, the impact limiter is assumed to absorb, in plastic deformation of Hexcel, the potential energy of the drop event. That is, the analyses assume that none of the drop potential energy is transferred to kinetic energy or strain energy of the target (the "unyielding surface" assumption of 10 CFR 71) nor strain energy in the vessel itself.

For each drop orientation, the prediction of overpack behavior can be approached from straight forward energy balance principles:

$$E = W(h+c) = \int_{0}^{\infty} Fx \, dx$$

Where: W = package weight

h = Drop height

- -

- c = maximum overpack deformation, or crushing
- Fx = Force imposed upon target and package by the overpack at a deflection equal to x.

The left-hand term represents the potential energy of the drop. The right-hand term represents the strain energy of the deformed overpack.

Figure 2.4 shows the tributary area of resistance at incremental depths of crushing. Using the effective area of resistance at each incremental depth of crushing, effective load-deformation (energy absorption) characteristics of the impact limiter have been developed for the two drop cases. These load-deformation characteristics for the two cases are shown in Figures 2.5 and 2.6, respectively. The detailed calculations are given in Appendices 2.10.1. The maximum depth of crushing, peak deceleration and reaction loads generated are shown on Table 2.8.

2.6.7.3 ANSYS - Three Dimensional Finite Element Stress Analysis

Detailed three-dimensional finite element models of the critical sections of the vessel were prepared using eight node threedimensional isoparametric solid elements (ANSYS element STIF 45). These models for the two cases are shown in Figures 2.7 and 2.8, respectively. For the side drop, a typical one-foot wide section of the vessel wall was modeled using three one-inch thick layers. The density of the vessel wall and gravel/grout (hereinafter, concrete) elements were selected to give proper distribution of weight of the vessel and that of the concrete and reactor vessel internals. The modulus of elasticity of concrete was conservatively taken as 28,000 psi (1/1000 of the modulus of steel) so that the concrete does not provide any stiffness to the reactor vessel wall. For the edge drop model, the vessel was modeled as two 1-1/2 inch thick layers. To account for the weakened section at nozzle openings, corresponding elements were removed. Appropriate boundary conditions were applied at the plane of symmetry.

At each incremental depth of crushing, appropriate boundary conditions and acceleration values were applied to depict the resistance offered by the impact limiter. For each model, ten load steps were applied at 1/2 inch increments in depth of crushing. Typical stress contours for primary membrane stress and membrane plus bending stress are shown in Figures 2.9 through 2.20. For the side drop cases the load is shown applied on the right side of the figure, and for the edge drop case the load is applied on the bottom of the figure. Maximum stresses for each incremental depth of crushing are summarized and compared against their allowable values in Table 2.9. The stresses in the vessel wall shown in Table 2.9 do not necessarily increase uniformly with increased depth of crushing. At lower depths of crushing, the effective resistance force and the width of support offered by the impact limiter is smaller. With lower resistance force the stress in the vessel wall is smaller. However, the smaller support width results in higher stress in the vessel wall. At increased depth of crushing the resistance force and effective support widths are larger so that the stress levels do not increase uniformly with depth of crushing. From these results, the margins of safety are shown in Table 2.10.

Results of finite element analyses were verified by hand calculations using analytical procedures given in Welding Research Council Bulletin WRC 107 (Ref. 2.4) which is widely used for the design of pressure vessels. The detailed design calculations for the impact limiter, finite element analysis results and WRC 107 calculations are given in Appendix 2.10.5.

2.6.8 Corner Drop

The corner drop evaluation, as specified in 10 CFR 71.71 (c)(8), applies only to fiberboard or wood rectangular packages not exceeding 110 lbs, and fiberboard or wooden cylindrical packages not exceeding 220 lbs. Therefore, the evaluation is not required for the Pathfinder package.

2.6.9 Compression

The compression evaluation as specified in 10 CFR 71.71 (C)(9), applies to packages weighing up to 11,000 lbs. Therefore, the evaluation is not required for the Pathfinder package.

2.6.10 Penetration

The penetration evaluation as specified in 10 CFR 71.71 (c)(10), requires an analysis of a 13 lb bar with a diameter of 1-1/4 in. dropped from a height of 40 in. onto the exposed surface of the

package which is expected to be most vulnerable to puncture. The long axis of the cylinder must be perpendicular to the package surface. For a 40 in. drop the impact velocity is 14.7 ft/sec. Using the Ballistic Research Lab (BRL) analysis (Ref. 2.6), the thickness plate that will be perforated is 0.015 in. Since the reactor vessel wall is 3 inches thick, and the nozzle closure plates are 3/4 inch thick the bar will not penetrate the vessel. Similarly, the external kinetic energy is 43.33 ft lbs. Using the Stanford Research Institute (SRI) analysis (Ref. 2.6), the thickness of the plate that will be perforated is 0.0039 in. Therefore, there is adequate margin of safety to prevent penetration of the bar through the plate. Results of the penetration analysis is shown in Appendix 2.10.6-B.

2.6.11 Load Resistance

Routinely, packages have been evaluated with respect to structural standards for "load resistance". Specifically, it is usually required to regard the package as a simple beam supported at its ends with a loading equal to five times its fully loaded weight. Detailed structural calculations are given in Appendix 2.10.6-C. The maximum stress is 0.72 ksi in the bolts at the closure head. The margin of safety is (26.81/0.72) - 1 = 36.2.

2.7 Hypothetical Accident Conditions

This evaluation is not applicable.

2.8 Special Form

This evaluation is not applicable.

2.9 Fuel Rods

This evaluation is not applicable.

Rev. 0

2.10 Appendix

Contents

2.10.1 Calculations of Package Weight and Center of Gravity

2.10.2 Reduced External Pressure Calculations and Finite Element Analysis

- 2.10.3 Increased External Pressure Calculations
- 2.10.4 Side Drop and Edge Drop Finite Element Analysis

2.10.5 Design Calculations for Impact Limiter, Finite Element Analysis, and WRC 107

2.10.6

- -A Brittle Fracture Evaluation
- -B Penetration Evaluation
- -C Load Resistance Calculations

2.10.7 References



APPENDIX 2.10.1

Calculations of Package Weight and Center of Gravity



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Length (FT)

*** VOID ***

*** volume of vessel ***

Printed on 14-Sep-89

Internals Gravel/Grout (CF) (LB)

Volume (CF)

weight Distribution for the Pathfinder vessel

CONTRACTOR CONTRACTOR

Component	TOTAL
··· Vessel ···	
Vessel Head	24029.500
Section 1	30546 450
Section 2	67354.650
Section 3	11 21540.200
Section 4	5935.050
Section 5	3136.320
section 6	1742.400
vessel Totals:	154284.570
*** internals ***	
Feedmater Ring Assembly	979.000
Hold Down Stud Assembly	213.000
Borter Polson Shim	483.000
Single Element Box Atta	10 000
triad Sample bolder	2464 000
Irrad Sample Container	42.000
Four Flement Cluster Box LESS	5200 000
Feedwart King Support Assy	48 000
Superhtr Control Rod Assy	975 000
Superhtr Control Rod voke	100.000
Steam Separ, Group Assy	\$ 175.000
Reactor Grid Plate	5600.000
Boiler Shroud Assembly	1460.000
Steam Sep. Support Shaft	2310.000
Superheaters Shell Assy	4800.000
Boiler Control Rods	2674.000
HOTO DOWN ASSENDIN	7000.000
ateas bryer	3400.000
Internal Totals:	42900.000
*** Shielding ***	
Interior cylinder	36764 000
Bottom Plate	3286.800
Outside Shell:	1
Top Plate	620.110
Outside shell Upper Section	1 1012.210
upper middle Plate	789.900
Outside shell middle Section	11243.261
Lower Middle Plate	1130.530
Cutside shell Lower Section	1017.040
Fill material	277.990
Top Section	380 676
Middle Section	6520 134
Bottom Section	258 807
Shielding totals	63310.02659
*** Gravel & Grout ***	
Section 1	
Section 2	199363 605
Section 3	43064 628
Section 4	6406 274
Section 5	2252 447
vessel Head	1 16388 619
Gravel & Grout Totals:	320081 158
TOTAL WEIGHT	
	580584.754

16.16

Section 1 - 130e 5-1301.5 Diameter (1) Diameter (2) Diameter (3) Diameter (4) Diameter (5) Diameter (6) Diameter (7)	1 000 7 560 7 904 8 592 9 260 9 968 10 656 11 000	23.461 53.461 62.747 72.776 83.548 46.046	0.543 1.065 1.321 1.558 1.665 0.775	3596 114 8215 719 9635 171 11171.252 12844 159 7101.169
Section 1 Total:		342.059	6 947	52565 563
Section 2 + 1301 5-1287.5 • Diameter	1.000	47.517 95.033	0.775 1.658 3.015 4.273 4.277 4.277 4.277 4.277 4.333 4.332 4.332 4.332 4.332 4.332 4.332 1.696	7331 429 14646 753 14434 025 14236 619 14236 021 14236 021 14236 021 14236 021 14227 201 14227 201 14227 381 14227 381 14227 381 14227 381 14227 181
Section 2 Total		1330 464	40 372	1 199 41 604
Section 3 - 1287.5-1282.7	1.000 5.500 5.438 5.375 4.188 3.625 2.125 2.000	47.045 92.348 73.449 48.709 28.258 3.142	1 009 12 507 3 343 0 493 0 030 0 000	7080.066 12523.661 10996.825 7563.199 4427.867 492.790
Section 3 Total		292.950	18 281	43084.046
Section 4 - 1282.75-1279. - Diameter	1.000	9.425 12.566 12.566 6.283	0.000 0.000 0.000 0.000	1478.371 1971.161 1971.161 1971.161 985.580
Section 4 Total		40.841	0.000	6406 . 273
Section 5 - 1279.5-1277.6 - Radius (1) - Radius (2) - Radius (3)	1.000 2.000 1.820 1.130 0.750	5 809 7 732 0 377 0 442	0.000 0.000 0.000 0.000	911 136 1212 913 59 099 59 299
Section 5 Total		14.360	000 0	2252 .447
Section 6 - Height - Diameter I Section 6 Total		n/a		
Vessel Head - 1310-125-13 - Radius (1) - Radius (2) - Radius (3) - Radius (4) - Radius (5)	1 000 0 900 2 700 3 430 3 710 3 760	0 160 13 247 30 455 40 624 22 098	0 000 0 000 0 477 1 085 0 543	25 108 2077 932 4702 274 6202 144 3381 161
Vessel Head Total		106.584	2.105	16388.619
TOTAL VOID		2127 .258	86.705	320061 158
Gravel & Grout Density (LB/C Internals Density (LB/CF) Density of Hexcel (LB/CF) Volume of Hexcel needed	90	156.86 495.00 5.00 1433.68		

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2.2-5 1 HUSANN DATE 9)13/89 SECTION 2.2 DATE 10/22/29 SUB-SECTION _____ OF CHKD . PROJECT NO. 4-228 TASK NO PHIS84 PAGE TITLE NSP - TATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Contain: CENTER OF GRAVITY CALCULATIONS with Origin at Topof Head EL. 1312.0' X= } 3.7209×0.5+5.6233×1.5+7.675×2.5+10.534×3.5 + 12 · 334×45+ 13 · 680×55+ 15 · 484×6.5+ 17 02×7.5 + 18.672×85+20.397×95+21.280×105+20.598×11.5 + 21.356×12.5+22,438×13.5+24.409×145+

24,409 ×15,5 + 24,409 ×16,5+24,409 ×17,5+24,43×185

+ 24,428×19.5 + 24.428×20.5 + 24.429×21.5

+ 24.424x22.5 + 26.00x23.5+24.045x24.5

+ 26.495×25.5+ 20.433×265+ 18.338×27.5

+ 9.836x 28.5+ 5.345 x 29.5+4,668 30.5

+ 4.668×31.5 + 4.514×32.5 + 3.752×33.5

+ 1.474×34.5+1.747×35.5} +

= <u>9911.68</u> = 17.0245'

APPENDIX 2.10.2

Reduced External Pressure Calculations and Finite Element Analysis


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I CRITERIA & METHOD OF APPROACH

The bounding case for this normal transport condition is specified in 10 CFR 71.71 (c) (3), which indicates a 3.5 psi absolute pressure.

The atmospheric pressure is 14.5 psi, the pressure across the shell wall would be 11.0 psi. Combining this pressure with the maximum internal pressure of \odot 0 psi that can develop during the hot environment condition results in a maximum pressure differential of 110 psi. The membrane hoop stress associated with this pressure is equal to pr/t, which for a \equiv in. thick and domin. radius cylinder, is $\mathbf{2}_1$ ksi. For an S_m allowable of 12.6 ksi, the resulting margin of safety is

M. S. = 12.6/0.242-1 = 51.07

NB-3324 Tentative Pressure Thickness

The following formulas are given as an aid to the designer for determining a tentative thickness for use in the design. They are not to be construed as formulas for acceptable thicknesses. However, except in local regions (NB-3221.2), the wall thickness of a vessel shall never be less than that obtained from the formulas in NB-3324.1 and NB-3324.2, in which:

t =thickness of shell or head, in. $= \frac{1}{2} \cdot 0^{1/2}$ P =Design Pressure, psi $= 11 \cdot 0$ R =inside radius of shell or head, in. $= 66 \cdot 0^{1/2}$ $R_o =$ outside radius of shell or head, in. $S_m =$ design stress intensity values (Tables I-1.0), psi = 1/2/600.0 psi

NB-3324.1 Cylindrical Shells

$$t = \frac{PR}{S_m - 0.5P} \quad \text{or} \quad t = \frac{PR_o}{S_m + 0.5P}$$

NB-3324.2 Spherical Shells

$$t = \frac{PR}{2S_m - 1} \quad \text{or} \quad t = \frac{PR_o}{2S_m}$$

$$Trey = \frac{11.0 \times 66.0}{12,600 - 0.5 \times 11.0}$$

= 0.0576"
- M.S = $\frac{3.0}{0.0576} - 1$
= 51.04 0.K



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	Reactor Vessel Analysis as Type A'shipping. Contains
RESULTS	OF FINITE ELEMENT ANALYSIS

**** POST-PROCESSING OF PFV1.SO

**** DUTPUT FOR TYPE-6 ELEMENTS (GROUP 1) LOAD CASE = 1 SCALE = 1.0000E+00

*** ALGEBRAIC MAXIMA

	TOP- SI	TOP- S2	51-52 (B)	S1-S2 (T)	MBRN 51-52	MEMBRANE S1
ELEM	79	79	78	259	75	78
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	BC	60	81	260	74	77
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	94	78	89	255	76	79
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	77	94	73	254	73	76
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	76	53	95	258	80	91
	2.7229E+01	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	93	3 77	75	261	96	92
	2.7229E+03	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	86	95	92	262	77	81
	2.7229E+02	1.756CE+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	83	76	91	256	79	73
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
ELEM	86	88	87	257	91	75
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.65636+02	2.4275E+02
ELEM	95	6 87	82	263	95	90
	2.7229E+02	1.7560E+02	2.3457E+02	1.1226E+02	1.6563E+02	2.4275E+02
						ALC REPAIR AND A DESCRIPTION OF A DESCRIPTION OF A DESCRIPTION OF A DESCRIPT

Map. Pormary Membranestories in vessel wall = 242.8 psr ALLowalde Stress = 12,600,0 Peri Margin of Safety M.S = 12600 -1 = 50.89

= 51



USALL DATE 10-16-89 SECTION 263-4 DATE 10/22/59 SUB-SECTION ____ OF . PROJECTING 4-228 TASK NO FHISBY PAGE TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A'shipping Container EVALUATION OF THE THE NOZZI STRESS AT COVER PLATE NOZZLE DIAMETER = 22.0" COVER PLATE THICKNESS = t= 0.75" Stockses in the cover plate due to poessure lond will be evaluated using Formulas for Flate Plate with fixed Boundary conditions R.J. ROARKS FORMULAS FORSTRESS & FROM STRAIN " 5th Edition, McGraw Hill Book Co.

TABLE 24 Formulas for flat circular plates of constant thickness

•

NOTATION: W = total applied load (pounds); v = unit line load (pounds per inch of circumference); q = load per unit area (pounds per square inch); $M_e = \text{unit applied line moment loading (inch-pounds per inch of circumference)}$; $\theta_e = \text{externally applied change in angular displacement (radians)}$; $y_e = \text{externally applied step in the normal displacement (inches)}$; y = vertical deflection of plate (inches); $\theta = \text{slope of plate measured from the horizontal (radians)}$; $M_e = \text{unit radial bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inch-pounds per inch of circumference})$; $M_i = \text{unit tangential bending moment (inches) per inch of circumference}$; $M_i = \text{unit tangential bending moment (inches)}$; r = radial bocation of expansion (inches) per inch of circumference; E = modulus of elasticity (pounds per square inch); i = Poisson's ratio; y = temperature coefficient of expansion (inches) per inch per degree; o = outer radius (inches); b = inner radius for annut plate (inches); i = plate thickness (inches); r = radial location of quantity being evaluated (inches); $r_e = \text{radial location of unit line loading or the start of a distributed load (inches)$. F_i to F_k and G_i to G_{2k} are the several functions of the radial location r. C_i to C_k are plate constants dependent upon the ratio a/b. L_i to L_{2k} are loading constants dependent upon the ratio a/r_e . When used as subs

Positive signs are associated with the several quantities in the following manner: Deflections y and y_e are positive upward; slopes θ and θ_e are positive when the deflection y increases positively as r increases: moments M_r , M_t , and M_e are positive when creating compression on the top surface; and the shear force Q is positive when acting upward on the inner edge of a given annular section

Bending stresses can be found from the moments M_i and M_i by the expression $\sigma = 6M/t^2$. The plate constant $D = Et^2/12(1 - r^2)$. The singularity function brackets () indicate that the expression contained within the brackets must be equated to zero unless $r > r_c$, after which they are treated as any other brackets.

Case no., kunding, kind terns	Edge testraint	Boundary values	
10. Uniformly distributed load from τ_{e} to e $T_{e} = \frac{-qr^{a}}{D}G_{11}$ $LT_{e} = \frac{-qr^{a}}{D}G_{1e}$ $LT_{e} = -qr^{2}G_{1e}$ $LT_{e} = -qr^{2}G_{1e}$ $LT_{e} = -qr^{2}G_{1e}$	10a. Simply supported	$y_{w} = 0 \qquad M_{w} = 0$ $y_{e} = \frac{-q\sigma^{4}}{2D} \left(\frac{L_{1}}{1+r} - 2L_{1} \right)$ $M_{e} = qe^{y}L_{1},$ $\theta_{e} = \frac{q}{8Da(1+r)} (a^{2} - r_{e}^{2})^{2}$ $Q_{e} = \frac{-q}{2e} (a^{2} - r_{e}^{2})$	$s = K_{*} \frac{ga^{*}}{D} \theta = K_{*} \frac{ga^{*}}{D} \qquad M = K_{*} ga^{*}$ $\frac{r_{*}/a}{r_{*}/a} = 0.063767 \qquad -0.05767 \qquad -0.04221 \qquad -0.02303 \qquad -0.00677$ $\frac{K_{*}}{K_{*}} = 0.06315 \qquad 0.75862 \qquad 0.06785 \qquad 0.03939 \qquad 0.0124$ $\frac{K_{*}}{K_{*}} = 0.2625 \qquad 0.17540 \qquad 0.11972 \qquad 0.06215 \qquad 0.01774$ $Nete \ 1(r_{*} = 0, \qquad G_{11} = \frac{1}{64} \qquad G_{14} = \frac{1}{16} \qquad G_{17} = \frac{(3+1)}{16}$ $y_{*} = \frac{-qa^{*}(5+v)}{64D(1+v)} \qquad M_{*} = \frac{qa^{2}(3+v)}{16} \qquad \theta_{*} = \frac{qa^{3}}{8D(1+v)}$
	10b. Fixed	$y_{\bullet} = 0 \qquad \theta_{\bullet} = 0$ $y_{e} = \frac{-\varphi a^{*}}{2D} (L_{1\bullet} - 2L_{1i})$ $S_{e} = \varphi a^{2}(1 + e)L_{1\bullet}$ $M_{ee} = \frac{-\varphi}{8a^{2}} (a^{2} - r_{e}^{2})^{2}$	$\begin{array}{c ccccccccccccccccccccccccccccccccccc$

HUSELLA DATE 10 16 89 SECTION _ DATE 10 22 16 SUB-SECTION ___ PROJECT NO NO.4-22 B TASK NO PHISP4 PAGE __ THE NSP- PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shitting Containe EVALUATION OF STRESS @ THE NOZZLE CONER PLATE (Contd.) - max moment Me = Wazer+ = - 9 (11:0) 2 (1+ 0.3) = 9.83 a nave. Moment Mra = - Nation = N(11.0)2 = 15.13 9 For 11.0 psia external Pressure Mayo Moment = 15.13×11.0 = 166.4 168in/i For 34" thick Plate, Max Bending Streeds = 116.4 × 6 = 1,775.3 psi

APPENDIX 2.10.3

Increased External Pressure Calculations

Hudding DATE 911/89 DATE 15 22 16 SUB-SECTION _____ PROJECT NO 4-228 TASK NOPHISPAPAGE TTLE NSP - PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shipping Container OBJECTIVE ! TO PERFORM ANALYSIS FOR IN CREASE EXTERNAL PRESSURE I CRITERIA & METHOD OF APPROACH The IDCFR71 subpart F. Paragraph 71.71(5) (4) requires evaluation for an external pressure equal to 20ps absolute. Allowable external pressures for ASME Code vessels are established by paragraph NB-3133 of ASME Code Section III (Ref. 2.3.)

NB-3133 Components Under External Pressure

NB-3133.1 General. Rules are given in this paragraph for determining the stresses under external pressure loading in spherical and cylindrical shells with or without stiffening rings, and tubular products consisting of pipes, tubes, and fittings. Charts for determining the stresses in shells, hemispherical heads, and tubular products are given in Appendix VII.

NB-3133.2 Nomenclature. The symbols used in this paragraph are defined as follows:

A = factor determined from Fig. VII-1100-1 in Appendix VII and used to enter the applicable material chart in Appendix VII. For the case of cylinders having D_c/T values less than 10, see NB-3133.3(b). Also, factor determined from the applicable chart in Appendix VII for the material used in a stiffening ring, corresponding to the factor B and the design metal temperature for the shell under consideration. (REF. 2.3)



- BY 1 HUSAND DATE 911 89 SECTION 216.4-2 CHKD AR DATE 10/22/85 SUBSECTION OF PROJECT NO 4-228 TASK NO PHISBA PAGE OF TITLE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A'Shipping Container
- A, = cross-sectional area of a stiffening ring, sq in.
- B=factor determined from the applicable chart in Appendix VII for the material used in a shell or stiffening ring at the design metal temperature, psi
- D_e=outside diameter of the cylindrical shell course or tube under consideration, in.
- E=modulus of elacticity of material at design temperature, psi (for this value, see Table I-6.0). Use the curve with this value on the material/temperature line of the applicable chart in Appendix VII.
- I = available moment of inertia of the combined ring-shell section about its neutral axis, parallel to the axis of the shell, in.⁴ The width of the shell which is taken as contributing to the combined moment of inertia shall not be greater than $1.10\sqrt{D_cT}$, and shall be taken as lying one-half on each side of the centroid of the ring. Portions of shell plates shall not be considered as contributing area to more than one stiffening ring.
- I_s= required moment of inertia of the combined ring-shell section about its neutral axis parallel to the axis of the shell, in.⁴
- L = total length, in., of a tube between tubesheets, or the design length of a vessel section, taken as the largest of the following:

(1) the distance between head tangent lines plus one-third of the depth of each head if there are no stiffering rings;

(2) the greatest center-to-center distance between any two adjacent stiffening rings; or

(3) the distance from the center of the first stiffening ring to the head tangent line plus one-third of the depth of the head, all measured parallel to the axis of the vessel, in.

L_i = one-half the distance, in., from the center line of the stiffening ring to the next line of support on one side, plus one-half of the center line distance to the next line of support on the other side of the stiffening ring, both measured parallel to the axis of the component. A line of support is:

(1) a stiffening ring that meets the requirements of this paragraph;

(2) a circumferential line on a head at one-third the depth of the head from the head tangent line; or

(3) circumferential connection to a jacket for a jacketed section of a cylindrial shell. P=external design pressure, psi (gage or absolute, as required)

- Pe=allowable external pressure, psi (gage a absolute, as required)
- R = inside radius of spherical shell, in.
- S= the lesser of 1.5 times the stress intensity at design metal temperature from Tables I-1.0 or 0.9 times the tabulated yield strength at design metal temperature from Tables I-2.0
- T = minimum required thickness of cylindrical shell or tube, or spherical shell, in.
- T_n == nominal thickness used, less corrosion allowance, of cylindrical shell or tube, in.

NB-3133.3 Cylindrical Shells and Tubular Products (a) The minimum thickness of cylindrical shells or tubular products under external pressure having D_e/T values equal to or greater then 10 shall be determined by the procedure given in Steps 1 through 8 below. Step 1: Assume a value for T. Determine the ratios

- Step 1: Assume a value for T. Determine the ratios L/D_e and D_e/T .
- Step 2: Enter Fig. VII-1100-1 in Appendix VII at the value of L/D_{o} determined in Step 1. For values of L/D_{o} greater than 50, enter the chart at a value of L/D_{o} of 50. For values of L/D_{o} less than 0.05, enter the chart at a value of L/D_{o} of 0.05.
- Step 3: Move horizontally to the line for the value of D_o/T determined in Step 1. Interpolation may be made for intermediate values of D_o/T . From this intersection move vertically downward and read the value of factor A.
- Step 4: Using the value of A calculated in Step 3. enter the applicable material chart in Appendix VII for the material/temperature under consideration. Move vertically to an intersection with the material/temperature line for the design temperature. Interpola tion may be made between lines for interme diate temperatures. In cases where the value at A falls to the right of the end of the material/temperature line, assume an inter section with the horizontal projection of the upper end of the material/temperature line For values of A falling to the left of the material line, see Step 7.

Step 5: From the intersection obtained in Step .

• ELES	BY DATE 9/1/89 SECTION 2.6.4-3 CHKD DATE 1922 85 SUB-SECTION OF PROJECT NO 9-228 TASK NO. PHISBY PAGE OF TITLE NSP · PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Container
2.6.4 Increase For Circular Cylin Step 1. For L:	d External Pressure (Contid.) nder: $T = 3.0''; D_0 = 2x69.0 = 138.0''$ $= (13x12+8) + \frac{1}{3}x69x2 = 210.0''$
:. L = :- L Do =	= (21x12 + 625) - 3625 + 690 = 2910 = (21x12 + 625) - 3625 + 630 = 2450 = (21x12 + 625) - 3625 + 630 = 2450 = 2910" 2910" $\frac{29}{2} = 2.109 \qquad (46.0)$
• <u>step 2</u> A = 0.0 5 B = 15,	4B 4x15,100.0
For the spherical	3Do/T 3x 46.0 437.7 pst 20.0 pst
$A = \frac{0.125}{R_{1T}}$ $B = 17.0$ $Pa = \frac{B}{R_{1T}}$	$= \frac{0.125\times5.0}{69.0} = 0.0054$ 00.0 form Figure VII-1100-2 $= \frac{17000\times3}{69.0} = 739.1482$
•	20.0 psi.



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Reac	tor vess	el Analysis as	: Type A'sh	ipping Contain

move horizontally to the right and read the value of B.

Step 6: Using this value of B, calculate the maximum allowable external pressure P_e using the following formula

$$P_{\bullet} = \frac{4B}{3D_{\bullet}/T}$$

Step 7: For values of A falling to the left of the applicable material/temperature line, the value of P_e can be calculated using the following formula:

$$P_{\bullet} = \frac{2AE}{3D_{\bullet}/T}$$

- kp 8: Compare P, with P. If P, is smaller than P, select a larger value for T and repeat the design procedure until a value for P, is obtained that is equal to or greater than P.
 1) The minimum thickness of cylindrical shells or the products under external pressure having D_e/T to less than 10 shall be determined by the modure given in Steps 1 through 4 below.
- in 1: Using the same procedure as given in (a) above, obtain the value of B. For values of D_c/T less than 4, the value of factor A can be calculated using the following formula:

$$A = \frac{1.1}{(D_{\bullet}/T)^2}$$

For values of A greater than 0.10 use a value of 0.10.

in 2: Using the value of B obtained in Step 1, calculate a value P_{e1} using the following formula:

$$P_{e1} = \left(\frac{2.167}{D_{e}/T} - 0.0833\right) B$$

Step 3: Criculate a value P_{e2} using the following formula:

$$P_{e1} = \frac{2S}{D_e/T} \left(1 - \frac{1}{D_e/T} \right)$$

Step 4: The smaller of the values of P_{e1} calculated in Step 2 or P_{e2} calculated ir Step 3 shall be used for the maximum allowable external pressure P_e . Compare P_e with P. If P_e is smaller than P, select a larger value for T and repeat the design procedure until a value for P_e is obtained that is equal to or greater than P.

NB-3133.4 Spherical Shells. The minimum required thickness of a spherical shell under external pressure, either seamless or of built-up construction with butt joints, shall be determined by the procedure given in Steps 1 through 6 below.

Step 1: Assume a value for T and calculate the value of factor A using the following formula:

$$A = \frac{0.125}{R/T} = 0.0054$$

- Step 2: Using the value of A calculated in Step 1, enter the applicable material chart in Appendix VII for the material under consideration. Move vertically to an intersection with the material/temperature line for the design temperature. Interpolation may be made between lines for intermediate temperatures. In cases where the value of A falls to the right of the end of the material/temperature line, assume an intersection with the horizontal projection of the upper end of the material/temperature line. For values at A falling to the left of the material/temperature line, see Step 5.
- Step 3: From the intersection obtained in Step 2, move horizontally to the right and read the value of factor $B_{.} = (7,000)$
- Step 4: Using the value of B obtained in Step 3, calculate the value of the maximum allowable external pressure P_e using the following formula:

$$P_{\bullet} = \frac{B}{R/T} = \frac{17000}{69.0} \times \frac{3}{5}$$

= 739.161

Step 5: For values of A falling to the left of the applicable material/temperature line for the Design Temperature, the value of P_e can be calculated using the following formula:

$$P_{\bullet} = \frac{0.0625E}{\left(R \cdot T\right)^{\prime}}$$



50.0

40.0

35.0

30.0 25.0

20.0 18.0 16.0

14.0 12.0 10.0

> 9.0 ...

7.0 6.0

50

4.0 3.5 3.0 2.8 2.0 1.8 16 1.4 1.2



FIG. VII-1100-1 GEOMETRIC CHART FOR CYLINDERS UNDER EXTERNAL PRESSURE OR COMPRESSIVE LOADINGS

Diam Outsude (e turth

0



Strength 30,000 to 38,000 psi, inclusive) AND TYPE 405 AND TYPE 410 STAINLESS STEEL



BY 1. Husan	DATE 91189 SECTION 2.6.4-1
CHKD	DATE 10/2.2/8 SUB-SECTION OF
PROJECT NO. 4-2	2.B TASK NO. PH15.84 PAGE OF
TITLE NSP . PATS	FINDER VESSEL SHIPPING ANALYSIS
Reactor V.	issel Analysis as Type A' Shipping Contain

Step 6: Compare P. obtained in Step 4 or 5 with P. If P. is smaller than P, select a larger value for T and repeat the design procedure until a value for P. is obtained that is equal to or greater than P.

NB-3133.5 Stiffening Rings for Cylindrical Shells (a) The required moment of inertia of the combined ring-shell section is given by the formula:

$$I_{*} = \frac{D_{*}^{3}L_{*}(T + A_{*}/L_{*})A}{10.9}$$

The available moment of inertia I for a stiffening ring shall be determined by the procedure given in Steps 1 through 6 below.

Step 1: Assuming that the shell has been designed and D_e , L_e , and T_e are known, select a member to be used for the stiffening ring and determine its area A and the value of Idefined in NB-3133.2. Then calculate B by the formula:

$$B = \frac{3}{4} \left(\frac{PD_0}{T_1 + A_1/L_1} \right)$$

- Step 2: Enter the right-hand side of the applicable material chart in Appendix VII for the material under consideration at the value of B determined in Step 1. If different materials are used for the shell and stiffening ring, then use the material chart resulting in the larger value for factor A in Steps 4 or 5 below.
- Step 3. Move horizontally to the left to the material/temperature line for the design metal temperature. For values of *B* falling below the left end of the material/temperature line, see Step 5.
- Step 4: Move vertically to the bottom of the chart and read the value of A.
- Step 5: For values of B falling below the left end of the material/temperature line for the design temperature, the value of A can be calculated using the following formula:

$$A = 2B/E$$

required I, is smaller than the computed I for the section selected in Step 1, that section should be satisfactory.

(b) Stiffening rings may be attached to either the outside or the inside of the component by continuous welding.

NB-3133.6 Cylinders Under Axial Compression. The maximum allowable compressive stress to be used in the design of cylindrical shells and tubular products subjected to loadings that produce longitudinal compressive stresses in the shell or wall shall be the lesser of the values given in (a) or (b) below:

(a) the S_m value for the applicable material at design temperature given in Table I-1.1 or Table I-1.2;

(b) the value of the factor B determined from the applicable chart contained in Appendix VII, using the following definitions for the symbols on the charts:

- T=minimum required thickness of the shell or tubular product, exclusive of the corrosion allowance, in.
- R = inside radius of the cylindrical shell or tubular product, in.

The value of B shall be determined from the applicable chart contained in Appendix VII as given in Steps 1 through 5 below.

Step 1: Using the selected values of T and R, calculate the value of factor A using the following formula:

$$A = \frac{0.125}{R/T}$$

- Step 2: Using the value of A calculated in Step 1, enter the applicable material chart in Appendix VII for the material under consideration. Move vertically to an intersection with the material/temperature line for the design temperature. Interpolation may be made between lines for intermediate temperatures. In cases where the value at A falls to the right of the end of the material/temperature line, assume an intersection with the horizontal projection of the upper end of the material/temperature line. For values of A falling to the left of the material/ temperature line, see Step 4.
- Step 3: From the intersection obtained in Step 2, move horizontally to the right and read the value of factor B. This is the maximum allowable compressive stress for the values of T and R used in Step 1.

Step 4: For values of A failing to the left of the



HUSAM DATE 10-16-89 SECTION 26.4-8 - DATE 10 22 15 SUB-SECTION ____ OF PROJECT NO. 4-228 TASK NO. FHISBA PAGE TITLE NSP . PATHEINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A'shipping Contained EVALUATION OF THE STRESS THE NOZZIF AT

CONER PLATE NOZZLE DIAMETER= 22.0" COVER PLATE THICKNESS = t= 0.75" Stockers in the cover plate due to Poessure Lond will be evaluated using Formulas for Flate Plate with fixed Boundary Conditions FROM R.J. ROARK'S FORMULAS FOR STRESS & STRAIN "5th Edition, McGraw Hill Book Co.

TABLE 24 Formulas for flat circular plates of constant thickness

NOTATION: W = total applied load (pounds); w = unit line load (pounds per inch of circumference); q = load per unit area (pounds per square inch); $M_{\phi} = \text{out applied line moment loading (inch-pounds per inch of circumference)}$; $\theta_{\phi} = \text{externally applied change in angular displacement (radians)}$; $y_{\phi} = \text{externally applied step in the normal displacement (inches)}$; y = vertical deflection of plate (inches); $\theta = \text{slope of plate measured from the horizontal (radians)}; M_{\phi} = \text{unit radial bending moment (inch-pounds per inch of circumference)}; M_{e} = \text{unit tangential bending moment (inch-pounds per inch of circumference)}; M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of circumference); M_{e} = unit tangential bending moment (inch-pounds per inch of radius); Q = unit shear force (pounds per inch of circumference); E = modulus of elasticity (pounds per square inch); r = Poisson's ratio; y = temperature coefficient of expansion (inches per inch of elegret); e = outer radius (inches); b = inner radius for annular plate (inches); t = plate thickness (inches); r = radial location of quantity being evaluated (inches); r_{e} = radial location of unit line loading or the start of a distributed load (inches). F_{1} to F_{2} and G_{1} to G_{26}$ are the several functions of the radial location r. G_{1} to G_{2} are plate constants dependent upon the ratio e/f_{e} . L_{1} to L_{20} are loading constants dependent upon the ratio e/f_{e} . When used as subscripts, r and t refer to an evalua

Positive signs are associated with the several quantities in the following manner: Deflection g and g_e are positive upward; slopes θ and θ_e are positive when the deflection g increases positively as r increases; moments M_{e} , M_{e} , and M_{e} are positive when creating compression on the top surface; and the shear force Q is positive when acting upward on the inner edge of a given annular section

Bending stresses can be found from the moments M, and M, by the expression $v = 6M/t^2$. The plate constant $D = Et^3/12(1 - v^2)$. The singularity function brackets () indicate that the expression contained within the brackets must be equated to zero unless $r > r_e$, after which they are treated as any other brackets.

Case no., knading, knad terms	Ealge restraint	Boundary values	
18. Uniformly electributed load from r_{e} to u $LT_{e} = \frac{-qr^{e}}{D}C_{11}$ $LT_{e} = -\frac{qr^{2}}{D}C_{12}$ $LT_{e} = -qr^{2}C_{12}$ $LT_{e} = -qr^{2}C_{12}$ $LT_{e} = -qr^{2}C_{12}$	10a. Supply supported	$y_{e} = 0 \qquad M_{e} = 0$ $y_{e} = \frac{-qe^{2}}{2D} \left(\frac{L_{11}}{1+r} - 2L_{11} \right)$ $M_{e} = qe^{2}L_{11}$ $\theta_{e} = \frac{q}{8Du((1+r)}(e^{2} - r_{e}^{2})^{2}$ $Q_{e} = \frac{-q}{2e}(e^{2} - r_{e}^{2})$	$\begin{cases} y = K_{+} \frac{qe^{2}}{D} & \theta = K_{+} \frac{qe^{2}}{D} & M = K_{+} qe^{2} \\ \hline r_{+} r_{-} & 0.0 & 0.0 & 0.0 & 0.4 & 0.6 & 0.8 \\ \hline r_{+} & -0.06370 & -0.05767 & -0.04221 & -0.02303 & -0.0007 \\ \hline K_{+} & 0.06615 & 0.05662 & 0.06785 & 0.03939 & 0.0124 \\ \hline K_{+} & 0.20625 & 0.17540 & 0.11972 & 0.00215 & 0.0177 \\ \hline \Lambda_{effe} & 1(r_{+} = 0) & G_{11} = \frac{1}{64} & G_{14} = \frac{1}{16} & G_{17} = \frac{(3+r)}{16} \\ \hline y_{+} = \frac{-qa^{4}(5+r)}{64D_{1}(1+r)} & M_{+} = \frac{qa^{3}r_{+} + r}{16} & \theta_{+} = \frac{qa^{2}}{8D_{1}(1+r)} \end{cases}$
	10b. Fixed	$y_{n} = 0 \theta_{n} = 0$ $y_{n} = \frac{-q a^{n}}{2D} (L_{14} - 2L_{11})$ $M_{n} = q a^{2} (1 + r) L_{14}$ $M_{n0} = \frac{-q}{8a^{2}} (a^{2} - r_{0}^{2})^{2}$	$\begin{aligned} \frac{r_{a}/u}{K_{a}} & 0.0 & 0.2 & 0.4 & 0.6 & 0.8 \\ \hline K_{a} & -0.01563 & -0.01336 & -0.00329 & -0.00334 & -0.0005 \\ K_{a} & 0.08125 & 0.06020 & 0.03152 & 0.01095 & 0.00150 \\ K_{a} & -0.12500 & -0.1520 & -0.05520 & -0.05120 & -0.01620 \\ \hline Neter II r_{a} = 0, & G_{11} = \frac{1}{64} & G_{14} = \frac{1}{16} & G_{17} = \frac{(3+r)}{16} \\ y_{1} = \frac{-qu^{4}}{64D} & M_{a} = \frac{(a^{2}(1+r))}{16} & M_{m} = \frac{-qu^{2}}{8} \end{aligned}$

DATE 10/16/89 SECTION 2.6. DATE 10 22 PS SUB-SECTION _ PROJECT NO. 10.4-22 B TASK NO. PH 1584 PAGE THE NSP - PATHFINDER DECOMISSIONING STUD Reactor Vessel Analysis - Type A Shitting Container,

EVALUATION OF STRESS @ THE NOZZLE COVER PLATE (Coutd.)

Max moment Me = Va2(1+12) = v(1110)2(1+0.3)

= 9.839

Mayo, Moment Mra = - 1/ az

= N(11.0)2 = 15.13 V

For Increased External Pressure of 20.0 pti. Max Moment = 15.113 × 20 = 302.6 Ibsin./in. For 34" Huck Cover Plate, Section Modulus = 5 = 1 (0.75)² in³/in Map. Bending Stoess = 302.6 × 6 I× (0.75)²

= 3,227.7 PSI

3,23 Kgi

APPENDIX 2.10.4

Side Drop and Edge Drop Finite Element Analysis



BY 1. Husam DATE 10/16/89 SECTION 2.6.7-1 CHKD _____ DATE _____ SUB-SECTION _____ OF ____ PROJECT NO 4-228 TASK NO PHISBA PAGE _____ OF ____ TITLE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Container I OBJECTIVE - TO PERFORM PATHFINDER VESSEL DROP ANALYSIS FOR DNE FOOT DROP.

+ ASSUMPTIONS: ASSUMPTIONS ARE LISTED IN COMPUTATIONS WHERE THEY OCCUR

I REFERENCES - GENERAL REFERENCES ARE GIVEN ON PAGE 36 OF THIS APPENDIX SPECIFIC REFERENCES ARE PROVIDED WHERE EVER THEY DECUR



EV 1 HULDOND DATE 10/16/89 SECTION 2.6.7-2 DATE 10/22/8 9 SUB-SECTION ____ OF PROJECT NO. 4-228 TASK NO FHISBA PAGE TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' shipping Contain: IN DESCRIPTION

2.6.7 Free Drop

The Pathfinder package weighs 291 tons. Subsection F of 10 CFR 71 requires that a package of this mass be analyzed for a drop of one foot onto a flat, essentially unyielding, horizontal surface, striking the surface in a position for which maximum damage is expected. During transport the package will always be oriented in the horizontal position; i.e., the centerline axis of the package will be horizontal. Because of the unique size, weight, mass and bulk of the package, no other orientation is considered feasible during normal transport conditions. Consequently, the free drop condition (which is expected to result in maximum damage) is a onefoot side drop of the package on an essentially unyielding horizontal surface.

Two cases were evaluated: (1) drop of the package along its length, and (2) impact of the package at its edge, wherein 50 per cent of the package weight and kinetic energy is transmitted to one edge of the package. Drop of the package at its lower edge was considered to be more critical as it is heavier due to the core internals

weight. Figures 1 and 2 on page 2.67-4, shows the orientation

of these two drop cases. of a heaviest section of the vessel and its internals along its length. It should be noted that for the vertical drop, the weight of the postruding parts of the package will be traismothed to the impact limiters at the edge of the vessel. (tubuch are analyzed for a signe freantly larger weight and kinetic energy in case 2). Therefore the maximum tributory (effective) weight for drop case I will be that of the heavest one foot unde section of the vessel and internals along its length.



4515

2.6.7.1 Analytical Methods

This section describes the analytical methodology used to demonstrate the structural adequacy of the package during a side or edge drop event. The analytical method is based on a step-by-step evaluation of external energy, resistance offered and energy absorbed by the impact limiter. The maximum stresses in the reactor vessel wall are evaluated at each incremental depth of crushing. The Section 2.6.7.2 describes the calculation of deformation and forces generated during the drop event. Section 2.6.7.3 describes the ANSYS three dimensional finite element models used for the detailed evaluation of maximum stresses in the package.

2.6.7.2 Impact Limiter Resistance/Deformation Behavior

As described in Section 1.2, the package includes a circular energy absorbing impact limiter surrounding the vessel. This impact limiter is designed to protect the package during a one-foot drop event. The impact limiter consists of precrushed Hexcel energy absorbing material. For purposes of analysis, the impact limiter is assumed to absorb, in plastic deformation of Hexcel, the potential energy of the drop event. That is, the analyses assume that none of the drop potential energy is transferred to kinetic energy or strain energy of the target (the "unyielding surface" assumption of 10 CFR 71) nor strain energy in the vessel itself.

For each drop orientation, the prediction of overpack behavior can be approached from straightforward energy balance principles:

 $E = W(h+c) = \int_{0}^{c} Fx dx$

Where: W = package weight

- h = Drop height
- maximum overpack deformation, or crushing
- I'x = Force imposed upon target and package by the overpack at a deflection equal to x.

The left-hand term represents the potential energy of the drop. The right-hand term represents the strain energy of the deformed overpack.

Figure 3 shows the tributary area of resistance at incremental depths of crushing. Using the effective area of resistance at each incremental depth of crushing, effective load-deformation (energy absorbtion) characteristics of the impact limiter have been developed for the two drop cases. These load-deformation characteristics for the two cases are shown in Figures 4 and 5 respec-

tively.





APPENDIX 2.10.5

Design Calculations for Impact Limiter, Finite Element Analysis, and WRC 107





DATE 10/16/29 SECTION 2.6.7er 1 Husan DATE 10/22/99 SUB-SECTION ____ OF __ OHKD . PROJECT NO 4-228 TASK NO PHIS84 PAGE OF ... TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Container CASEI Max. Acceleration from VIZ=ViZ+Zas. Vt = 0 ; impact velocity Vi = 8.02 ft/sec s. depth of conshing = 5.17 in · acceleration a = $\frac{V_1^2}{25} = \frac{(8.02)^2 \times 12}{2 \times 5.17} = 2.32g$ Total Vestical 6 mod = 24.5 (1+2.32) = 81.4 K 4121.22K FIGURE 3 HEXCEL PROPERTIES AND WIDTH AND DEPTH OF CRUSHING





BY 1. HUSAN DATE 10/16/89 SECTION 2.6.7-8 122 DATE 10/22/89 SUB-SECTION ____ OF ___ CHKD _ PROJECT NO 4-228 TASK NO PHISBA PAGE OF TITLE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS SIDE DROP (COSE) Vessel Analysis as Type A'shipping Container Using the above Procedure, for various depth of coushing, o and maximum resistance Load WOR

Depth of crushing (m)	0 (°) ·	Max. Resistance Lond (K)
0	0	0
0.2	.5°	2.9.65
- 0.5	6.46	383
1.0	9.14	54.04
1.2	100.	59.1
1.5	11.2	66.1
2.0	13.0	76.2
2.5	14.5	850
2.68	15	88.1
3.0	15.9	93.0
3.5	17.2	1003
4.0	18:3	107.0
4.5	19.5	113.4
. 4.75	20	116.4
5.0	20.53	119.3
5.11	20.8	121.2



BY 1. HUADING DATE 9121 189 SECTION 21617-9
CHKD DATE DATE SUB-SECTION OF
ENNE PROJECT NO. 4-665 TASK NO. THISP'T PAGE OF
CASEZ. Reactor Vessel Analysis as Type A' Shipping Contruin
DESIGN OF HEXCEL CORE FOR SIDE DROP AT ENCE
DROP HEIGHT = 1'= 12' TOTAL WEIGHT = 582.2 K
CONSIDER 50% OF TOTAL WEIGHT DROP@EDGE
MAX. VELOCITY OF IMPACT = VI = B.021 Holes
EXTERNAL KINETIC ENERGY OF IMPACT L
Exi = 582.2x0.5x12 = 3,493.2 jn.K
FOR HEXCEL 4-5052- 004 7.9 Pcf. 1
CRUSHING STRENGTH for= 725 psi (Page 2.6.7-12)
Total External Energy with HEXCEL Deformation of 5.18"= 582 ? (12+5.18)
For 3×12 = 36" wide Cylinder = Ex
1 - 1 - 172 75)2/2-5001.84
Volume of Hexcel Crushed = = = (10.15) (0-5100) -0=V
: Internal Energy Absorbed by HEXCEL CORE
$E_1 = V fe = \pm (78.75)^2 (0 - sino) 36 \times \frac{725}{1000}$
= 80,930.4 (0-sino) in.K
Equating Internal & External Energy
$80,930.4(0-sin0) = \frac{582.2}{2}(12+5.18) = 5001.1$
0 - 5i u 0 = 0.0618
$0 = 41.52^{\circ}$
:- Depth of Conshing = h= R(1-cos 2) = 78.75(1-cos 41.52) = 5.11" 2 5.18" O.K.
% of Total Thekness = 5.11 = 0,639 20.7 D.K.
병사가 집에 가지 않는 것은 것이 같이 있다. 것은 것을 가지 않는 것을 수 있는 것을 하는 것을 수 있는 것을 하는 것을 수 있는 것을 하는 것을 수 있는 것을 하는 것을 하는 것을 하는 것을 하는 것을 하는 것을 수 있는 것을 하는 것을 하는 것을 하는 것을 수 있는 것을 하는 것을 수 있는 것을 것을 것을 것을 것을 것을 것을 수 있는 것을

CASE Z.	A-ZZB TASK PATHFINDER V DATE VESSEL AND	9)21/89 SECTION 10/22/PS SUB-SECTION NO FHISE4 PAGE - ESSEL SHIPP NYSIS AS TYME	DN <u>2.6.7-10</u> DTION OF
Design of HEXCEL (ORE FOR	SIDE DRO	PONEDGE
Max. width $C = 2RS$ = 5 Max. REACTION LO Max. REACTION LO	5:83" > 5:83" > $AD DEVEL = 1,457,1 \frac{V_{12}}{259} = (8)$	x78.75 Si 48'' O.K Usedivithe OPED = 56 072.8 140 = $102.)^2 x 12$ $5.02.)^2 x 12$	andysis 5.83 ×36×725 1,457.1K = 2.35G
Total Load = 582.2x Using the Above Proce Case are Depth of Crushing (in) h	(1+2.35) = dure, Lond . Q()	970.7 K -Deformation <u>max. Re</u> Full Model	21,457.1 K m data for this sistance Lond (K) Symmetrical model
0.3	5°	358.4	179.2
10	6.46	4628	231.4
1.2	9.14	6 53.2	32616
1.5	10	714.0	357.
	11.2	7 98.9	399.4
2.0	13.0	920.8	4-60.4
215	14.5	1028.0	514.0
2.68	15	1064.0	532.0
3.0	15.9	11240	561.4
3.5	17.2	12120	6 06.0
4.0	18.3	1243.0	644.5
4.5	19.5	1370.0	6 851
4.75	20	1406.0	703
5.0	20.53	1442.0	720.8
5.11	20.8	1457.0	28.6



tlo	BY). HUBALLA DATE S116189 SECTION 2.6.7-12 DATE 10/22/89 SUBSECTION CF
	TITLE NSP - PATHFINDER DECOMISSION ING STUDY REACTOR Vessel Avalusis - TUDE A Shibbing Container
From H	excel Catelog (Reference 212, 2.7)

5052 ALLOY HEXAGONAL ALUMINUM HONEYCOMB

SPECIFICATION GRADE

MEXCEL	ALL N	COMPRESSIVE				1							
HONEYCOMB	8-	Bara			Stabiliza		1.	"L" Direction			***		
Cell-Material-Gage	1	Strer pi	hoth H	Stre	noth Bi	Modulus Nai	3	Streng	pth	Modulus	Stee	ngth li	Modulus
		typ	min	Mp	min	typical	typ	typ	min	typical	typ	min	typical
1/16-50520007	6.3	850°	-	900*	-	2750	-	6004	480*	909	375*	300*	400
1/16-5052001	9.0	16005	-	4700*	-	420*		900*	6350	105+	5450	520*	531
1/8-50520007	3.1	205	200	360	215	75	130	210	186	43.0	130	**	32.0
1/8-5052001	4.5	660	376	\$70	405	150	340	340	285	70.0	220		31.0
1/8-30520015	6.1	1000	660	1040		240	450	560	465	98.0	320	272	41.0
1/8-3052002	8.1	1485	1000	4615	1100	350	750	725	670	135	485	400	54.0
1/8-5052003	12.0	2800	2100	3050	2200	9009	-	1940*	1250	-	14300	10:00	
5/32-50520007	2.6	220	150	260	160	55	90	165	120	37.0	100	70	19.0
5/32-5052001	3.8	395	285	410	300	110	185	270	215	56.0	175	125	26.4
5/32-50520015	5.3	690	096	720	535	195	340	420	370	84.0	270	215	36.0
5/32-5052002	6.9	1060	770	1130	800	285	\$75	590	540	114	375	328	46.4
5/32-50520025	8.4	1530	1070	1600	1180	370	800	760	690	140	475	420	56.0
3/16-50520007	2.0	160	90	475	100	34	60	120	80	27.0	70	46	14.3
3/16-5052001	3.1	315	200	335	215	75	130	210	155	45.0	130		21.0
3/16-50520015	4.4		360	890	385	145	250	330	280	68.0	215	160	30.0
3/16-5052002	8.7	820	-	860	600	220	390	440	410	90.0	300	244	38.5
3/16-50520025	6.9	1120	770	1175	800	285	575	590	540	114	375	328	46.4
3/16-5052003	8.1	1000	1000	1720	1100	350	750	725	670	135	455	400	54.0
1/4-50520007	1.6	90	60	100	70	20	40	85	60	21.0	50	32	11.0
1/4-5052001	2.3	210	120	230	130	45	75	140	100	32.0	85	\$7	16.2
1/4-50520015	3.4	376	240	385	250		150	235	180	\$0.0	150	105	24.0
1/4-5052002	4.3	800	350	540	370	140	230	320	265	66.0	210	155	29.8
1/4-50520025	5.2	720	500	760	510	190	335	410	360	82.0	265	200	35.4
1/4-5052003	60	1050	630	1100	660	235	430	495	445	96.0	315	265	40.5
1/4-5052004	7.9	1420	970	1490	1050	340	725	700	650	130	440	390	\$2.0
3/8-50520007	1.0	45	20	50	20	10	25	45	32	12.0	30	20	7.0
3/8-5052001	1.6	90	60	95	70	20	.0	85	60	21.0	50	32	11.0
3/8-50520015	2.3	190	120	200	130	45	75	140	100	32.0	85	57	16.2
3/8-5052002	3.0	300	190	315	200	70	120	200	145	43.0	125	85	21.2
3/8-50520025	3.7	405	270	420	285	105	180	260	200	55.0	170	115	26.0
3/8-5052003	4.2	620	335	640	355	135	220	310	255	65.0	200	150	29.0
3/8-5052004	154	810	500	850	535	200	360	430	380	65.0	280	228	36.8
3/8-5052-005		1000	200	1	760	265	505	545	500	105	350	300	43.5

Votes:

1. Test data obtained at 0.625 inch thickness

2 p-Preliminaries (see page 11)

3 s-Predicted values

4. Hearn shear the 12.0 pct product.

Additional information on Hexcel is found on Pages 30 though 35.



BY 1. HUDDIN DATE 10/16/29 SECTION 3.6.7-13 CHKD DATE 10/22/89 SUB-SECTION OF ____ PRO :ECT NO 4 - 228_ TASK NO PHISBA PAGE _____ OF _____ Reactor Vessel Analysis as Type A' Shipping Container

MAXIMUM STRESS SUMMARY

	Sid	de Drop	Ed	Edge Drop				
Depth of Crushing, in	Max. Membrane Stress, ksi	Max. Membrane + Bending, ksi	Max. Membrane Stress, ksi	Max. Membrane + Bending, ksi				
0.5	2.56	5.33	3.03	13.06				
1.0	2.74	5.54	3.79	13.45				
1.5	3.35	6.77	4.63	16.44				
2.0	1.91	6.25	4.49	17.11				
2.5	2.13	6.97	5.01	19.10				
3.0	2.33	7.63	5.47	19.27				
3.5	2.52	8.23	5.90	20.78				
4.0	2.39	7.27	5.00	20.26				
4.5	2.54	7.77	5.30	21.47				
5.11	2.71	8.23	5.63	22.83				

MEMBRANE AND MEMBRANE PLUS BENDING MARGINS OF SAFETY

Maximum membrane stress	-	5.90	ksi
Allowable membrane stress = Sm		17.88	ksi
Margin of safety = 17.88/5.90 - 1	=	2.03	
Membrane plus bending stress	=	22.83	ksi
Allowable membrane plus bending = 1.5 Sm		26.81	ksi
Margin of safety = 26.81/ 22.83 - 1	-	0.17	7
Maximum tensile stress in liosure head bolts	=	0.51	ksi
Allowable tensile stress in bolts = Sm	=	17.88	ksi
Margin of safety = 17.88/0.31 -1	=	34.06	
Maximum shear stress in bolts	=	0.43	ksi
Allowable shear stress in the bolts = 0.6 X 38.0) =	22.80	ksi
Margin of safety = 22.8/0.43 -1	-	52.02	

BY 1. HUBAN DATE 10/16/89 SECTION 2.6.7-14 DATE 10/22/84 SUB-SECTION ____ OF ___ CHKD PROJECT NO. 4-228 TASK NO PHISBA PAGE _ OF . TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Container WELDING RESEARCH COUNCIL BULLETIN 107 (WRE 107 ENALVSIC Pot. 7.6 Results of finite element analyses were verified by hand calculations using analytical procedures given in Welding Research Council Bulletin WRC 107 (Ref. 2.4) which is widely used for the design of reactor vessels. De-se Calculations and curves used in the colculations are given on pages 15 through 24 From these results it can be seen that For Case 1 maximum streets in Cylindrical Vessel wall due to radial Lord of 108 K = 6.83 KS1. :- For Mano. Resistance Lond of 121.2 K Map. Membrane + Bending Stoess (Stores Intensity) = 6.83×121:2 = 7.66 KSr Compared to ANSYS Analytis result of 8.23 KSI. For Case 2 -Max stress For Spherical shell = 0.98 KSi for Radial Load of 1000K 11 11 $'' = 0.48 \times 1457.0 = 14.28 \text{ KSL}$ As compare to ANSYS Analysis result of 22.83 KSi. WRE 107 result for Case 2 are lower because they are for spherical shell rather thom junctuon of spherical and aphindrical shall



	BY 1. Husann	DATE 0/25/89	SECTION 216.7-15
	PROJECT NO. NO . 4 - 2	2B TASK NO. PHISPA	AGE OF
	TITLE NSP - PATHE Reactor Vess	INDER DECOMISSIO	NING STUDY R A Shipping Container.
WRC107 EL	ALUA TION	FOR SIDE	E DROP(CASEI)
PARAMETERS:	" NOZZEL	DIMENSION	US CRUTTERE
tor ie as	· · · · · · · · · · · · · · · · · · ·	~"	-
C.Z.=	6, 6, -		
PI = ST	$=\frac{10}{70.5}=0.2$	55	
B2 - C2 -	6 = 0.085	-1	
Fm	70.5		
$\frac{\beta_1}{\beta_2} = \frac{18}{6} =$	3.0 >1		
B=FI=+ (B)	-1) (1- K)] .	E.B.	
E No LIS B2		FIFZ	
$for m \phi, K = 0.0$	11	7 525	- 0.139
$\therefore P = \begin{bmatrix} 1 - \frac{1}{2} \times 1 \end{bmatrix}$	(5-1) (1-0.4]	[]] [- 0101
For Nx , K= 1.	68		
$B = \begin{bmatrix} 1 - \frac{2}{3} \end{bmatrix}$	(1-1.68)] 0.	1474 = 0.214	-2
For Ma, KI= 1.7	6		
:- B = [1- 3	(1-1.76)]0,14	.74 = 0.2221	
For Mar, KI = 1.	2 7	-1/71	
$\therefore \beta = \left[1 - \frac{2}{3}\right]$. (1-1.2) 0.147	4 = 0116/1	



								,P	xcc	-2= 56 x
I. Applie Red Cin Lon To She	f Lood 3" tal too f, t. Manuni, g. Biomonis, cian Maxani, ar Lood, ar Lood,			23.5	5		200 ×		ATTA	CHMENT
1. Game	try sal thickness, schwont rodius,		ending loss	ed. Kn = -	- Rm			2.2	BL (0)
Ina	Rand curren	Compare absolute values of	STRESS			eite that i			shown	
Pip	ter	Attess and anter results .	Au	41	Bu	BL	C.	CL	Du	DL
36	P/A 6	A. (P/A.) A.T 3 x70.5	193	-	-	-				
16	== = = = = = = =	" (+)····································	-54	1.	-	1	-		-	
34	H4/ Ralf =	Ra (Ho Raifi) · Maifit					-	-	<u> ·</u>	·
14	105	Rb (Mc/ Rm ft) + dMc .					-			-
38	HU	Ka (Ho) . ML .	-							
11.3		AL (MAR) . AML .	-						\sim	
	traically for summer		6.83			1929				
46		1. (m) 3.5x10t	-1.79	-	-		-	-	-	-
	M0. 0/6	K. (Ma.) . 4 . 0066 x6x106	47			1.	-		-	1.
	Ha	(+/ TI -75)2		285411	235400	S.S.C.M.	-	-		
	Me/Rmtfl	Me 'Amifi/ AmifiT	-	X	X					-
	He Roff	Me/Raff Refits	All Mars			Salanse	8900	38411	138///	188000
	ML/Am*#	(HL Antf) RatBT		-	·	·	\bigotimes			
28-1	ML/Raft	www.a.A. a.AT	-		•	-		anna.		
of X etres			6.54							
to Terais	m, Mr	rds . rad . 2017	+ .	•	•	•	•	•	•	•
Shoer str		140 - 3 VI		•	-	-				
Sharer atra		1.0 - VL						-	•	•
Add Algel	braically for summer					- New				
COMBINI		NTY, S								
1)		1 - K [a4 + 0. + Ving - 0. + + +	6.83						•	
		's - largest at 145. 0 0 0 - 0.	6.83		1.6.6				1	

 $N_{l}/(M_{L}/R_{n}^{1}\beta)$ so determined by (C_L) from Table

3 (see para. 4.3). 4.2.2.5.2: When considering bending moment $(M_i): \beta = K_L \sqrt[3]{\beta_1\beta_1}$ where K_L is given in Table 8.

10

4.3 Calculation of Stresses

4.3.1 STRESSES RESULTING FROM RADIAL LOAD, P.

4.3.1.1 Circumferential Stresses (*.):

Step 1. Using the applicable values of β and γ

Streases in Shells

Ref. 2.4

Checked by Dott 13/21/49

	Table						P.			2
UX =	<u>C1</u>	= = 24 PT = 0.875	J70.5x3.0		M		A PARTY	No.		GID
	U.C. IS	Z 10				AU	1	F	DL	1
Applied I Radio Sheer Sheer Overt	Loods* I Lood, Lood, Lood, urning Momern,	100.0 1	10		Con Star	ALK		An	TAL	咩
Terst	enel Memont,	4. Sure 1.	ancontrolion Factors		1	anna	MI IN	mu		
Y	ol Thickness, ol Moon Redius, hmont Outside Red	1	Enter all force values nee with sign conventi		Rm	SPI	HERICA	L SHE		
lalarance Fig. Nos.	Rend curves	Colculate obsolute values of stress and anter result s	STRESSES	- 11 100	d is oppos Bu	BL.	Cu	CL	Du	DL
18-2	HT = 023	$\mathbf{k}_{*}\left(\frac{\mathbf{H}_{*}\mathbf{T}}{T}\right)\cdot\frac{1}{T}=\frac{100 \times 107}{(3.0)^{2}}$	5 -0.26			-	-	-	-	-
Ţ		K. (+) . + = D. D. O. 08×6	X100 -0.72		-	•	-	•	-	·
18-3	1.T. R.T.	R. (HINT \ R. T) . B. T					-	-	•	•
	Hay Rat	R. (Mai, R.T.) . 44						•	•	-
	Hat Rat .	$R_{n}\left(\frac{M_{0}T\sqrt{R_{m}T}}{\mu_{1}}\right) = \frac{M_{2}}{T^{2}\sqrt{R_{m}T}}$	-	-		•]		<u>]</u>
+	Hay Rat .	$Eb \left(\frac{Be \sqrt{Bm}T}{B_2}\right) \cdot \frac{BM_2}{T 2 \sqrt{Bm}T} =$	-	•						
Add algebra of radial str	ically for summition	n. *	0.976							
SR - 2	MT =0.007	$\mathbf{I}_{\bullet} \left(\begin{array}{c} \mathbf{M}_{\bullet}^{T} \\ \mathbf{I}_{\bullet} \end{array} \right) \cdot \begin{array}{c} \mathbf{I}_{\bullet} \\ \mathbf{I}_{\bullet} \end{array} = \begin{array}{c} \mathbf{O}_{\bullet} \mathbf{O}_{\bullet}^{D} \\ \mathbf{I}_{\bullet} \end{array}$	-D.078	-	-	-	-		•	
+	# 20,0033	R. (H) . 1 . 0.0033x	200 -0.22		-	+	-	·	·	
se - 3	H.T. R.T.	$R_{n} \left(\frac{H_{Y}T_{1}/R_{m}T}{H_{1}}\right) \cdot \frac{H_{1}}{T^{2}\sqrt{R_{m}T}} =$]		-	-	•	·
	My \ RaT -	$K = \left(\frac{M_{T} \sqrt{R_{m}} T}{M_{T}} \right) \cdot \frac{\delta M_{T}}{T^{2} \sqrt{R_{m}} T} =$					-	•	•	-
	HyT (RmT	Kn (HyT\ RmT) + H2 "	-	-				3		·
	Mr. R.T	$Rb\left(\frac{M_{T_1}}{N_{T_1}}\right)\cdot\frac{\delta M_1}{T^{2}\sqrt{R_mT}}$	• *			-				
Add sigebri of tengantic	aically for summittee	n _y =	0.298							-
Sheer street		1 VI -					-	-	ŀ	
Sheer stre to load, Y	** ***	11	•		1	-				
Sheer stress to Tarstan,	is due MT	1. 1. 1. MT - T.	•	•			-		•	•
Add algobe	steally for summetter	•								
		i r , 1								
	Then # & #y he	" 1= " [". + ", · V 0,1"		•						1_
	like signs:	All should be a set of the set of	and the second se	No. of Concession, Name						

and radial bending stresses given in Figs. SR-2 and SR-3, so that the following simplified procedure for calculating maximum stresses has been developed.

DIAL LOAD (P).

Step 1. Calculate the value of the applicable shell parameter (U) as given in para. 3.2.1. Step 2. Enter Fig. SR-1 at the value of U found

3.6.1 MAXIMUM STRESS RESULTING FROM RA-

Ref. 2.4 CHECKED BY JAK 10/22/05

Stresses in Shells

6



2












2189 Sam DATE 91 SECTION Z DATE 10 22 84 SUB-SECTION ____ PROJECT NO 4-228 TASK NO PHISBY PAGE THE NSP - PATHFINDER VESSEL SHIPPING ANIAL 2:4] Reactor Vessel Analysis as Type A' shipping COMPUTATIONS - INFORMATION

V

M

σ.

Local Stresses in Spherical and Cylindrical Shells due to External Loadings

1. Nomenclature

.

S

K.

K.

Ø

E

P

Symbols used in the formulas and equations in the text are listed below:

1.1 General Nomenclature

- normal stress in the ith direction on the surface of the shell, psi
 - shear stress on the ith face of the jth direction
- stress intensity = twice maximum shear stress, psi

N_i = membrane force per unit length in the *i*th direction, lb/in.
 M_i = bending moment per unit length in

 bending moment per unit length in the *i*th direction, in. lb/in.

 membrane stress concentration factor (pure tension or compression)

- bending stress concentration factor
 denotes direction. In the case of spherical shells, this will refer to the tangential and radial directions with respect to an axis normal to the shell through the center of the attachment as shown in Fig. 1. In the case of cylindrical shells, this will refer to longitudinal and circumferential directions with respect to the axis of the cylinder as shown in Fig. 2.
- = denotes tensile stress (when associated with σ_i)
 - denotes compressive stress (when associated with σ_i)
 - = angle around attachment, degrees (see Figs. 1 and 2)
- = modulus of elasticity, psi

 concentrated radial load or total distributed radial load, lb

•	concentrated	external	shear	load,
-	concentrated	external	overt	

- moment, in lb
- Mr = concentrated external torsional moment, in. lb
- d., d. = inside diameter and mean diameter, respectively, of the nozzle, in.
- $D_{i}, D_{m} =$ inside diameter and mean diameter, respectively, of the shell, in.
- 1.2 Nomenclature Applicable to Spherical Shells

V,	 concentrated external shear load in 2-2 direction. lb
V	 concentrated external shear load in 1-1 direction, lb
Mı	 external overturning moment in 1-1 direction, in. lb
М,	 external overturning moment in 2-2 direction, in. lb
R.	= mean radius of spherical shell, in.
T	= thickness of spherical shell, in.
	- outside radius of outlindrical at
	tachment, in.
r.,	= mean radius of hollow cylindrical at-
	tachment, in.
	= thickness of hollow cylindrical et.
	tachment, in.
T	$= r_{\rm m}/t$
P	= T/t
II	- P/VRT
N' N'	- ro/ · rim1
N., N,	memorane force in shell wall in radial and circumferential direc- tions, respectively, lb/in. (see Fig. 1)
М., М,	 bending moment in shell wall in radial and circumferential direc- tions, respectively, in. lb/in. (see Fig. 1)
	= normal stress in radial direction nei

 normal stress in radial direction, psi (see Fig. 1)

 normal stress in circumferential direction, psi (see Fig. 1)

K. R. WICHMAN is with the Navy Dept., Bureau of Ships. Washington, D. C. A. G. HOPPER is with All States Design and Development Co., Inc., Pittaburgh, Pn. J. L. MERSHON is with the Atomic Energy Comnuminan, Washington, D. C. "The publication of this report is appropriate by the Pressure Vessel Re-

The publication of this report is sponsored by the Pressure Vessel Research Committee of the Welding Research Council.



NULSALLY DATE 913/89 SECTION 216.7-26 PROJECT NO 4-228 TASK NO PHISB4 PAGE _____ OF ____ TILE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A'shipping Containe

4. Cylindrical Shells

4.1 Sign Convention

Stresses will be considered in the shell at the attachment-to-shell juncture in both the circumferential and longitudinal directions as shown in Fig. 2. A knowledge of the shell deflections resulting from various modes of loading permits one to predict whether resulting stresses will be tensile (+) or compressive (-).



Consider Case I showing a direct radial inward load, P. Here P acts similar to a local external pressure on the shell causing compressive membrane stresses. Furthermore, local bending occurs so that tensile bending stresses result on the inside of the vessel at C and D while compressive bending stresses result on the outside.

In Cases II and III, the applied moments are considered to act as couples composed of equal and opposite radial forces. Hence, tensile membrane stresses result at B and D while compressive membrane stresses result at A and C. As in Case I



Fig. 2-Types of loading conditions at an attachment to a cylindrical shell

 $\left(\frac{\sigma_* T^* \sqrt{R_* T}}{M}\right) \left(\frac{M}{T^* \sqrt{R_* T}}\right)$

In using this formula, the sign conventions established in Table 1 should be used.

Ref. 2.4







	LOCATION			
-	22	:	1 Same	:
4.4		•		PK ISS
-				1
ħ	1944	1	1	Sand State
	i.	1		
Ħ		:		1990 A

tensile bending stresses result at A and C on the inside of the vessel, and B and D on the outside of the vessel. Similarly, compressive bending stresses result at A and C on the outside and B and D on the inside.

In this manner Table 4 has been developed to show the signs of stresses resulting from various external loading conditions. These stresses are located in the vessel wall at its juncture with the attachment. Use of Table 4 permits one to use the nondimensional curves presented in the following procedure with a minimum of encumbency and concern for sign convention.

The numerous stress components can be readily accounted for if a scheme similar to that shown in Table 5 is adopted. In using this scheme it is to be noted'that the Maximum Shear Theory has been used to determine equivalent stress intensities. Also it is to be noted that evaluation of stresses resulting from internal pressure has been omitted.

Test work conducted by PVRC has shown that stresses attenuate rapidly at points removed from the attachment-to-shell juncture, the maximum stress usually being located at the juncture. However, in the general case of arbitrary loading, one has no assurance that the absolute maximum stress intensity will be located at one of the eight points considered in the above discussion. The maximum stress intensity could be located at some intermediate point around the juncture under an arbitrary load, or under a longitudinal moment with the circumstances outlined in para. 4.4 and

Appendix A.

The results of Bijlaard's work have been plotted in terms of nondimensional geometric parameters by use of an electronic computer. Hence, the first step in this procedure is to evaluate the applicable geometric parameters γ and β .

4.2.1 SHELL PARAMETER (γ) . The shell parameter is given by the ratio of the shell mid-radius to shell thickness thus:

$$\gamma = R_{*}/T$$

4.2.2 ATTACHMENT PARAMETER (β). For cylindrical shells, either round or rectangular attachments may be considered in the following manner:

4.2.2.1 Round Attachment: For a round attachment the parameter β is evaluated using the expression:

$$\beta = \frac{0.875 r_{e}}{R_{e}}$$

4.2.2.2 Square Attachment: For a square attachment the parameter is evaluated by:

$$\beta = \beta_1 = \beta_2 = \frac{c_1}{R_{\pm}} \approx \frac{c_2}{R_{\pm}}$$

4.2.2.3 Rectangular Attachment Subject to Radial Load (P): For this case β is evaluated as follows:

$$\beta_1 = \frac{c_1}{R_{\rm m}} \qquad \beta_1 = \frac{c_2}{R_{\rm m}}$$

If
$$\frac{\beta_1}{\beta_1} > 1, \beta = \begin{bmatrix} 1 & -\frac{1}{3} \begin{pmatrix} \beta_1 \\ \beta_2 \end{pmatrix} - 1 \end{pmatrix} \begin{pmatrix} 1 & -K_1 \end{pmatrix} \end{bmatrix} \sqrt{\beta_1 \beta_1}$$

If $\frac{\beta_1}{\beta_1} < 1, \beta = \begin{bmatrix} 1 & 0 \\ 0 & 0 \end{bmatrix}$

$$\left[1-\frac{4}{3}\left(1-\frac{\beta_1}{\beta_1}\right)\left(1-K_1\right)\right]\sqrt{\beta_1\beta_1}$$

where K values are obtained from Table 6.

4.2.2.4 Rectangular Attachment Subject to Circumferential Moment (M.).

4.2.2.4.1: When considering membrane forces (N_i) : $\beta = \sqrt[3]{\beta_1^*} \frac{\beta_2}{\beta_2}$. Then multiply values of $N_i/(M_i/R_*\beta)$ so determined by C, from Table 7 (see para. 4.3).

4.2.2.4.2: When considering bending moment $(M_i): \beta = K_i \forall \beta_i \beta_i$, where K_i is given in Table 7.

4.2.2.5 Rectangular Attachment Subject to Longitudinal Moment (M_L).

4.2.2.5.1: When considering membrane forces (N_i) : $\beta = \sqrt[4]{\beta_1 \beta_2}$. Then multiply values of



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	Table 6-	-Radial Los	d (P)	
	N.	N.	M.	M,
K.	0.91	1.68	1.76	1.2
Ki	1.48	1.2	0.88	1.25

NOTE: Above holds approximately within limits 4 2 B./A. 2 1/0.

calculated in para. 4.2, enter Fig. 3C and read off the dimensionless membrane force $[N_{\bullet}/P/R_{\bullet}]$.

Step 2. By the same procedure used in Step 1, enter Fig. 1C or 1C-1 and find the dimensionless bending moment $[M_{\bullet}/P]$.

Step 3. Using applicable values of P, R, and T, find the circumferential membrane stress (N./T) by:

$$\frac{N_{\bullet}}{T} = \left[\frac{N_{\bullet}}{P/R_{\bullet}}\right] \cdot \left[\frac{P}{R_{\bullet}T}\right]$$

Step 4. By a procedure similar to that used in Step 3, find the circumferential bending stress (6M./T2) thus:

$$\frac{6M_{\bullet}}{T^{*}} = \begin{bmatrix} \frac{M_{\bullet}}{P} \end{bmatrix} \cdot \begin{bmatrix} \frac{6P}{T^{*}} \end{bmatrix}$$

Step 5. Combine the circumferential membrane and bending stresses by use of the general stress

equation (Section 2), together with the proper choice of sign (Table 4); i.e.: 4.2.4

4.3.1.2 Longitudinal Stresses (e,): Follow the 5 steps outlined in 4.3.1.1 except that $(N_*/P/R_*)$ is obtained using Fig. 4C; and [M, /P], using Fig. 2C or 2C-1. It follows that:

$$\frac{N_{r}}{T} = \left[\frac{N_{r}}{P/R_{\bullet}}\right] \cdot \left[\frac{P}{R_{\bullet}T}\right],$$

$$\frac{6M_{r}}{T^{*}} = \left[\frac{M_{r}}{P}\right] \cdot \left[\frac{6P}{T^{*}}\right] \text{ and }$$

$$\sigma_{r} = K_{\bullet} \frac{N_{r}}{T} = K_{\bullet} \frac{6M_{s}}{T^{*}}$$

4.3.2 STRESSES RESULTING FROM CIRCUMPEREN-TIAL MOMENT, M.

4.3.2.1 Circumferential Stresses (e.):

Step 1. Using the applicable values of β and γ calculated in pars. 4.2, enter Fig. 3A and read off the dimensionless membrane force $N_{\bullet}/(M_{\bullet}/R_{\bullet}^{*}\beta)$.

Step 2. By the same procedure used in Step 1, enter Fig. 1A and find the dimensionless bending moment $M_{\bullet}/(M_{\bullet}/R_{\bullet}\beta)$. Step 3. Using applicable values of M_{\bullet} , R_{\bullet} , β

		Table 1	Circumferential	Moment (M.)		
8./8.		K. for 0	K. for M.	K. for Ma	C. for No .	C. for M.
17.	16	1.09	1.31	1.84	0.31	0.49
	50	1.04	1.24	1.62	0.21	0.46
	100	0.97	1.16	1.45	.0.16	0.44
	300	0.92	1.02	1.17	. 0.09	0.46
14	15	1.00	1.09	1.36	0.64	0.75
	50	. 0.98	1.08	1.31	0.67	0.75
	100	0.94	1.04	1.26	0.51	0.76
	300	0.95	0.99	1.13	0.39	0.77
	1	(1.00)	(1.20)	(0.97)	(1.7)	(1.3)
	100	1.19	1.10	0.95	1.43	1.12
	300		(1.00)	(0.90)	(1.3)	(1.00)
14.011.041	15	(1.00)	(1.47)	(1.08)	(1.75)	(1.31)
	100	1 49	1.38	1.06	1.49	0.81
	300		.(1.27)	(0.98)	(1.36)	(0.74)

Nork: The values in parenthesis determined by an approximate solution.

							and the second se
-			Tabit	-Longitudinal M	oment (ML)		
	Ø1/Ø1	· 1	K . for e	Ki for Me	KL for M.	CL for No	C. for N.
		50	1.13	1.65	1.16	0.77	0.33
	1/.	300 15	1.31	1.56	1.11 (1.04)	0.90 (0.90)	0.07 (0.76)
	11	100	1.00	1.07 (1.05)	1.02 (1.02)	0.97 (1.10)	0.68 (0.60)
	2	15 ·	1.09	(0.94) 0.89	(1.12) 1.07	(0.87) 0.81	(1.30) 1.16
		300	- 1.39	(0.79)	(0.90)	(0,80) 0,68	(1.50)
		100 300	1.18	0.81 (0.64)	1.12 (0.83)	0.51 (0.60)	1.03 (1.33)

Nore: The values in parenthesis determined by an approximate solution.





C

Fig. 1-Types of loading conditions at an attachment to a spherical shell

In Case II, the overturning moment may be considered to act as a couple composed of equal and opposite radial forces. Hence, tensile membrane stresses result at A while compressive membrane stresses result at B. As in Case I, local bending also occurs so that tensile bending stresses develop at A on the outside of the vessel and at B on the inside, while compressive bending stresses develop at A on the inside and B on the outside.

In this manner, the signs (tensile +, compressive -) of stresses resulting from various external loading conditions may be predicted. It is to be noted that these stresses are located in the vessel wall at its juncture with the attachment.

Table 1 shows the signs of stresses resulting from radial load and overturning moment. This table will facilitate the use of the nondimensional curves (presented in the following procedure) and minimize concern for the signs of the calculated stresses.

3.2 Parameters

The results of Bijlaard's work have been plotted in terms of nondimensional geometric parameters by use of an electronic computer. Hence, the first step in this procedure is to evaluate the applicable geometric parameters.

3.2.1 SHELL PARAMETER (U). The shell parameter is given by the ratio of the nozzle outside radius to the square root of the product of shell radius and thickness, thus:

$$U = \frac{r_{\bullet}}{\sqrt{R_{\bullet}T}}$$

If a square attachment is to be considered, U may be approximated as follows:

3.2.2 ATTACHMENT PARAMETERS. For spherical shells, either round or square attachments may be considered.

3.2.2.1 Rigid Attachments: In the case of a rigid attachment, no attachment parameter is required to use the nondimensional curves.

3.2.2.2 Nozzles: For a hollow cylindrical attachment such as a nozzle, the following parameters must be evaluated:

$$T = \frac{r_0}{t}$$
$$r = \frac{T}{t}$$

3.2.2.3 Hollow Square Attachment: If a hollow square attachment such as a box beam is to be considered, the required parameters may be approximated as follows:

$$T = \frac{r_{-}}{0.875t}$$
$$p = \frac{T}{t}$$

3.3 Calculation of Stresses

3.3.1 STRESSES RESULTING FROM RADIAL LOAD, P



ROUND ATTACHMENT

SQUARE ATTACHMENT



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HEXCE	- CORE D	ETIGN - ADDI	TIONAL INFOR-
<u>(VIA)</u>	ON FROM	MBOLS	TRLOG
A	In ^a	Impact Area	Kets. 2.2 2.7

SYMBOLS

18.

A	ln²	Impact Area
a	Ft/sec ^a	Acceleration or Deceleration Raie
F	Lbs.	Impact Force
	PSI	Honeycomb Crush Strength
G	Dimensionless	Ratio a/g
a	Ft/sec ^a	Acceleration due to Gravity
KE	In-lb.	Kinetic Energy
m	Slugs	Mass
S	ln.	Stopping or Falling Distance
L.	ln.	Honeycomb Core Thickness
	Sec.	Time
v.,	Ft./Sec.	Initial Velocity
v.	Ft/sec.	Final Velocity
v	Ft/sec.	Impact Velocity-Often equal to ve
w	Lbs.	Impact Weight
e.	Lbs./Ft	Honeycomb Core Density

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FORMULAS

-, 2. 2, 2.7

The following formulas are those commonly encountered in simple energy absorption calculations.⁽¹⁾ General: ⁽²⁾

- 1. Kinetic Energy, $KE = \frac{1}{2} mv^2$
- 2. Mass, m = W/g
- 3. Dynamic Force, F = ma
- 5. Velocity, $v_t^2 = v_0^2 + 2\alpha S$

4. G == a/g

6. Distance travelled, $S = v_{et} + \frac{1}{2} at^2$

Honeycomb Energy Absorption:

7. Stopping Distance, S. From 5. above for v_f equal to impact velocity and v_e equal to zero, $v^2 = 2aS$

and from 4. above, a := Gg. Therefore: $S = \frac{v^2}{2aG}$

 Minimum core thickness. t. Assuming 70% of the total honeycomb thickness is available for crushing, then S = 0.7t, and therefore, from 7.

$$t_{c} = \frac{1}{0.7} \frac{\mathbf{v}^{*}}{2\sigma G}$$

This formula has been presented graphically on Page 3 of this bulletin.

9. Crush strength. fer, Since fer = F/A and from 3. above:

$$F = ma = \frac{Wa}{g} = WG$$

then
$$f_{er} = \frac{WG}{A}$$

This formula has been presented graphically on Page 4 of this bulletin.

It can be shown that:

$$KE = f_{cr} AS$$

from 1. above

$$KE = \frac{1}{2} mv^2 = \frac{wv^2}{2g} = \frac{WGS}{2g} = WGS = FS = I_{er} AS$$

Setting S equal to $0.7t_{\rm e}$ and solving for $t_{\rm e}$ will produce the final check equation presented on Page 3.

1. Formulas based on acceleration being a constant

2 g = 32.2 ft/sec² for earth environmer.; only





(2) See Page 12 for the solution to problems in which G is not a design consideration.



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EINETIC ENERGY EQUATION SOLUTION



This chart offers a rapid solution to the equation (1):

$$\mathbf{T} = \frac{1}{2}\mathbf{m}\mathbf{v}^2 = \frac{1}{2} \quad \frac{\mathbf{W}\mathbf{v}^2}{\mathbf{q}}$$



KINETIC ENERGY_EE_FOOT POUNDS



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KINETIC ENERGY ABSORPTION

This chart offers a solution to energy absorption problems in which so limiting G value is required (i.e. stop a payload in a specific distance with no rebound - - - G not critical). The chart graphically solves the equation:

KE = Le AS = L. A 0.7L

USE OF THIS CHART

- Enter the chart with the known value of KE in foot-pounds and total core thickness t, in inches. The intersection of the coordinates KE and t, defines a diagonal line.
- 2. Move along this diagonal to the intersection of the horizontal line representing the impact area A being considered.
- 3. At this point on the diagonal, move vertically upward on the chart and read the crush evength for required.



CORE THICENESS - L - NCHES

Section Section

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	PROJECT NO. 4-22 B TASK NO. PHISE PAGE OF
	THE NSP - PATHFINDER DECOMISSIONING STUDY
	Reactor Vessel Analysis - Type A Shipping Container

The load and deflection characteristics encountered in statically crushing honeycomb are graphically shown in the figure below. This curve indicates the value of the resistance to loading offered by the honeycomb as the compressive load is overcome and crushing starts and continues. Three items on this curve are of special interest:

> A. COMPRESSIVE PEAK - Honeycomb systems which are not precrushed or in which the initial contact area has not been reduced exhibit this peak force level at impact. This undesirable peak can be easily eliminated by proper design.

ef .2.7

- B. MAXIMUM CRUSH LEVEL -- A horizontal line has been drawn across this curve representing a maximum crush level. If this hypothetical line had been developed as a design maximum, then the actual honeycomb fer value must remain equal to or below this level for crushing to take place.
- C. MINIMUM STOPPING DISTANCE The vertical line drawn down this graph represents a hypothetical stopping distance minimum established in conjunction with the maximum crush level value. This value represents a minimum value for the maximum crush level line and any crush level selected below the maximum limit will require an increase in this minimum thickness.

	- PEAK
AD	1 or MAX.
10	r, MIK.
	DEFLECTION

"The actual crush strength of honeycomb is normally given as the average value of the load diagram inflections



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2.10.9 References

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- 2.4 Welding Research Council Bulletin No. 107, August 1985, "Local Stresses in Spherical and Cylindrical Shells Due to External Laodings:, by K.R. Wichman, et.al.
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- 2.8 "Pathfinder Atomic Power Plant Reactor Vessel Materials, Fabrication and Inspection", ACNP-62025, Allis-Chalmers Corporation December 1, 1962
- 2.9 "ASME Boiler and Pressure Vessel Code, Section III, Division 1, Subsection NB



APPENDIX 2.10.6-A Brittle Fracture Analysis



BY 1 HUSAUN DATE 9 3 89 SECTION 2 3.4-1 DATE 10 2489 SUB-SECTION ____ OF ____ PROJECT NO 4-228 TASK NO PHISBA PAGE TITLE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS OBJECTIVE: TO PERFORM BRITTLE FRACTURE EVALUATION

TI METHOD OF APPROACH

Bottle foncture evaluation will be performed using the procedure described in NUREG/CR -1815

UCRL-53013, RT " Recommendation for Protecting Against Failure by Bostlle Fracture in Ferritic Steel shipping containers up to four inches Tuck"

NOMENCLATURE

Ref. 2.5

- Temperature relative to NDT; A = LST TNDT
- AAR Association of American Railroads
- AWS American Welding Society B
 - Section thickness

B

A dimensionless parameter =
$$\frac{1}{B} \left(\frac{K_{ID}}{\sigma_{yd}} \right)^2$$

- COD Crack Opening Displacement
- CVN. Cv Charpy V-notch test or the test results
- DWTT Drop Weight Tear Test
- DT Dynamic Tear test or the test results
- HAZ Heat affected zone of welds
- KI Stress intensity factor
- KIC Critical value of K1 for static loading rates. When K1C is exceeded, fracture occurs.
- KID Critical value of K1 for dynamic loading rates
- KI(t) Critical value of K, for intermediate loading rates.
- LST Lowest service temperature (lowest metal temperature)
- (L) Limit of plane strain NDT
- Nil Ductility Transition O&T
- Quenched and Tempered 0
- Nominal stress (see glossary)
- Yield strength for a static loading rate. This is considered the ASTM minimum yield for a O'ys specific steel. 0 yd
 - Yield strength for dynamic loading rate
 - = σ_{ys} + 30 ksi for steels with $\sigma_{ys} \le 60$ ksi
 - = σ_{ys} + 15 ksi for steels with $\sigma_{ys} \ge 70$ ksi
 - = σ_{ys} + 20 ksi for steels with 60 ksi < σ_{ys} < 70 ksi
- Nil Ductility Transition (NDT) temperature TNDT (YC)
 - Yield Criterion; the level of toughness required to provide fracture arrest at a nominal stress equal to the yield strength.



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PROJECT NO. NO.4-	22B TASK NO PH 1584 PAGE OF
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Asp	er NUREGICE-1815, +	he Pathfind	er Reactor Vestel
للخبا	be evaluated to meet	t the foncture.	taughness requirements
ŧ	qualification procedures	atto cisted	with categoing
III	and Category II		Ref. 2.5

	Category		Calingary 1	Category II		Category III		
	Required degree of safety (see Appendix C)	Very large morges of safety.		Large margin of miety.		Adequate margin of soloty.		
	Required amount of fracture toughness (new Sec. 5).	Sufficient a teading : get	o arresi large erseks under dynamic noral yielding will precede fracture.	Sufficient to provent fracture initiation of pre-existing cracks under dynamic loading.		Sufficient to prevent fracture initiation at minor defects typical of good labrication practices.		
I		Thickness (B) (m.)	Annus Crittoria far meering raughneus (in.) requirements*		Criteria for meeting soughness requirements*	Thickness (B) (m.)	Critoria for monting taughnoon requirements	
		E	*NDT temperature ⁸ must be ion that a maximum value. See Fig. 3 and Seen. 5.1.1. 5.1.2. Add. if $a_{yy} \ge 70$ has, either: *5/8 is, there DT^4 must be greater than 400 P-th at upper shelf temperatures. See Sec. 5.1.1. E *Cv ⁴ must be greater than 45 ft-th at upper shelf temperatures. See Sec. 5.1.1.	-	*With full dynamic loading, NDT temperature ⁴ must be loss then a maximum value. See Fig. a and Sec. 5.2.1. Qr *With reduced loading rates, NDT temperature can be determined from Fig. 7. See Sec. 5.2.2.	-	*Without costing, are sormalized atom made to "Plue Grave Practice" or batter. Gr there that: *NDT ⁴ \leq 10°F (for 3 > 6.625 m.) Or tool to above that: *DT ⁴ > 50 ft.d. at 10°F, with 0.625 is, tool spectmen. Or cost to above that: *Cv ⁴ > 15 ft.d. at 10°F. Or *Without costing, are so-radied steel.	
		3	"Use DT Test E-404-80. 80% or greater stear fracture required at LST. See Sec. 5.1.3. Qr "Use DWTT Test E-436. 80% or greater shear fracture required	A STA	"Use DT Tex E-404-80. 50% or prester share fracture required of LST. See Sec. 5.2.3. Or "Use DWTT Tex E-434. 50% or greater share fracture required		provided wolds have been streap relieved and imperced by nondestructive evaluation techniques. See Sec. 5.3.1.	
			et LST. See Sec. 5.1.3.		at LST. See Sec. 5.2.3. Qr "Use say surmalized stori made to "File Grain Practice" ar better. See Sec. 5.2.3.	-	"Ne requirements when B is less than 8.4 in. thick. See Sec. 5.3.2.	
		A P	*Use Nurch Tennie Test E-338. Notch tennie strength yield strength > 1.0 at LST.	1	"No requirements when B is less than 0.19 in. See Sec. 5.2.3.			

"Full scale destructive testing on a case-by-case basis may be used as an alternate to requirements listed below.

^bNDT is measured according to ASTM E-208, or an equivalent NDT can be established by subtracting 50°F from the mide 5/8 in. DT energy transition curve measured according to ASTM E-604. at of the

¹NDT is measured according to ASTM E-208, or an equivalent NDT can be established by subtracting 50°F from the midpoint of the 5/K in. DT subtracting transition curve measured according to ASTM E-604, or the NDT temperature requirement can be met by milecting the maximum NDT temperature given in Fig. 1 or Table 3.

The measured according to ASTM F-604.



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53

5.3 CATEGORY III

The level of safety required for Category III is less than that for Category II. For Category III systems, fracture toughness must be sufficient to prevent fracture initiation at minor defects typical of good fabrication practices. A summary of fracture toughness requirements for Category III steels is shown in Table 6.

Good engineering practices and careful selection of the steel make it reasonable to expect that brittle fracture is unlikely to occur.

TABLE 6.	Category	III fracture toughness	requirement	and criteria	for ferritic steels	with vield	strength a	-
greater than	100 ksi.						au cugin i	

Required degree of safety (see Appendix C)	Adequate margin of safety.				
Required amount of fracture toughness (see Soc. 5)	Sufficient to prevent fracture initiation at minor defects typical of good fabrication practices.				
Thickness (B) (in.)	Criteria for meeting toughness requirements				
0.4 to 4.0	* Without testing, use any normalized steel made to "Fine Grain Practice" or better ⁸ . See Sec. 5.3.1.				
	Or				
	* Show that NDT ^b < 10*F (B > 0.625 in.).				
	Or				
	* Test to show that DT ^C > 50 ft-lb at 10°F, with test specimen 0.625 in. thick.				
	Or				
	* Test to show that Cyd > 15 ft-lb at 10°.				
	Or				
	* Without testing, use as-rolled steel, provided the welds have been stress relieved and imspected by mondestructive evaluation techniques.				
Less than 0.4	* No requirements when B is less than 0.4 in. See Sec. 5.3.2.				

*Steel with an NDT temperature lower tha. steels made to a fine grain practice.

^bNDT is measured according to ASTM E-208, or an equivalent NDT can be established by subtracting 50°F from the midpoint of the 5/8 in. DT energy transition curve measured according to ASTM E-604 or the NDT temperature requirement can be met by selecting the maximum NDT temperature given in Fig. 1 or Table 3.

^CDT measured according to ASTM E-604, for specimen thickness of 0.625 in.

Cy measured according to ASTM E-23.



BY _____ DATE _____ SECTION _____ 23.4-4 CHKD ______ DATE _____ DATE _____ OF _____ PROJECT NO. NO. 4-22B TASK NO. PHISE4 PAGE ______ OF _____ TITLE NSP- PATHFINDER DECOMISSION ING STUDY REACTOR VESSEL AVAILYSIS - TYPE A Shipping Container.

From Pathfindes Atomic Power Plant Report No. ACNP-62025, "Reactor Vessel Materials Fabrication and Inspection", by Allis-Chalmers Mar Co. Mil wankee, WI, December 1, 1962, Tol. 2.8

			TADLE II				
F10.		Haterial	N.P. Test No.	Yield Strength (PSI)	Tonsi is Strongth LPSI)	f ion- pet ion	Charpy V-Notch Impact Emergy at IO F (FT-Lb)
2	2	ASTN & 105-11	202	67,600 57,200	85,900 81,600	27.8 31.0	119 - 120 - 108 82 - 65 - 70
2	3	ASTN A212-0	200	42,900	71,190 71,400	×	17 - 16 - 15
2	2	ASTN A2 12-8	200	46 , 500	73,500 72,500	31	31 - 29 - 30
*	•	ASTN A2 12-0	212	42,500	72,600 75,000	32	19 - 15 - 17
2	•	ASTH A2 12-8	214	41,000	74,200 71,600	ĸ	21 - 20 - 16
2	5	ASTN A2 12-0	201	47,000	71,800 73,000	30	26 - 30 - 31
2	6	ASTH A 105-11	203	47,150 45,150	74,250 74,500	23.5 23	28 - 27 - 31 27 - 23 - 29
2	7	ASTM A2 12-8	2 12	40.500	76,500 74,900	м	16 - 16 - 17
2	8	ASTH A2 12-8	211	40,800	74,700 75,000	23	15 - 16 - 15
3	2	ASTH A2 12-P	248	44 , 900	76,100 76,600	32	19 - 23 - 18
3	3	ASTM A 105-11	247	46,700 45,300	70,000 71,500	40.5 36.5	110 - 140 - 126 136 - 115 - 68
•		ASTH A437 Gr 840	311-1	110,000	146 . 500 146 . 500	14	16 Ave (See Hote 2)
4		ASTH A437 G- 848	311-2	115,000	155,000	13 14	15 AVE (See Note 2)
5		ASTH A437 Gr C4C	222	103,750	125,000	20	32 - 28 - 30 (See Note 3)
•		ASTM A437 Gr 040	968	101,250	126, 900	20	35 - 34 - 32 (See Hote 3)
	Fig. 2 2 2 2 2 2 2 2 2 2 2 2 2	Fig. 11mm NO. 2 2 2 2 2 2 3 2 2 3 2 2 4 2 2 4 2 2 5 2 2 6 2 2 6 2 3 2 3 3 3 3 4 5 5	Fig. 1 tem NO. Motorial 2 2 ASTH A105-11 2 3 ASTH A105-11 2 3 ASTH A105-11 2 3 ASTH A212-6 2 2 ASTH A212-6 2 2 ASTH A212-6 2 4 ASTH A212-6 2 4 ASTH A212-7 2 5 ASTH A212-7 2 6 ASTH A212-7 2 6 ASTH A212-7 2 6 ASTH A212-7 2 6 ASTH A212-7 3 2 ASTH A212-7 3 2 ASTH A212-7 3 3 ASTH A212-7 3 3	Fig. item N.P. Ro. Ho. Heterisi Teet No. 2 2 ASTH A105-11 202 2 3 ASTH A105-11 202 2 3 ASTH A212-0 200 2 2 ASTH A212-0 200 2 2 ASTH A212-0 200 2 2 ASTH A212-0 212 2 4 ASTH A212-0 212 2 4 ASTH A212-0 214 2 5 ASTH A212-0 201 2 6 ASTH A212-0 201 2 6 ASTH A212-0 212 2 7 ASTH A212-0 212 2 7 ASTH A212-0 212 2 7 ASTH A212-0 212 2 8 ASTH A212-0 212 2 8 ASTH A212-0 211 3 2 ASTH A212-0 211 3 2 ASTH A437 311-1 4 ASTH A437 311-2 5 ASTH A437 333 6 ASTH A437 368 5 ASTH A437 968	TABLE 11 Yield Strength (Fig. 11em No: No: Heterial Yield Teet No. Yield Strength (FS1) 2 2 ASTM A105-11 202 67,600 57,200 2 3 ASTM A212-8 200 42,900 2 2 ASTM A212-8 200 42,900 2 2 ASTM A212-8 200 46,500 2 4 ASTM A212-8 212 42,500 2 4 ASTM A212-8 214 41,000 2 5 ASTM A212-8 201 47,000 2 6 ASTM A212-8 212 40,500 2 7 ASTM A212-8 212 40,500 2 7 ASTM A212-8 212 40,500 2 7 ASTM A212-8 211 40,800 3 2 ASTM A212-8 211 40,800 3 2 ASTM A212-8 211 40,800 3 3 ASTM A212-7 246 44,900 3 3 ASTM A437 311-1 10,000 4 ASTM A43	TABLE 11 Fig. ites N.P. Yisid Strength Strength 2 2 ASTM A105-11 202 67,600 65,900 2 3 ASTM A212+6 200 42,900 71,160 2 3 ASTM A212+6 200 42,900 71,160 2 2 ASTM A212+6 200 44,500 75,500 2 2 ASTM A212+6 212 42,500 72,600 2 4 ASTM A212+6 212 42,500 72,600 2 4 ASTM A212+6 214 41,000 74,200 2 4 ASTM A212+6 201 47,000 71,600 2 5 ASTM A212+6 201 47,000 74,200 2 5 ASTM A212+6 212 40,500 74,250 2 6 ASTM A212+6 212 40,500 74,700 2 7 ASTM A212+6 211 40,600	TABLE 11 Yield Tensile S Fig. Ho: Hoterial N.P., Test No. Yield Strength Strength

NOTESI

1. Keyho is impact at -50 F.

- 2. Charpy T-Notch Impacts at room temperature.
- 3. The Ispects at roos temperature.
- 4. Approved deviation from specified 30 Ft-Lb. Ave. Impact energy.
- 5. Approved deviation from specified tensils strength of 120,000 psi.





By 1. Husan	A DATE 916/89 SECTION	2.3.4-5
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PROJECT NO. 4-	228 TASK NO PHISBA PAGE	OF
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Reactor '	Vessel Analysis as Type	A' shipping Container

5.2 CATEGORY II

A summary of fracture toughness requirements for Category II steels is shown in Table 5.

The level of performance with respect to fracture safety necessary for Category II is less than that required for Category I. For Category II systems. fracture toug ness must be sufficient to prevent fracture initiation of pre-existing cracks under dynamic loading.

With Caregory II steels, fracture toughness (or the corresponding temperature relative to NDT) must be greater than that needed to exceed plane strain conditions ($\beta = 0.4$) under dynamic loading. We recommend that in defining the Category II

toughness requirement a value of $\beta = 0.6$ be used in the equation:

$$\theta = \frac{1}{B} \left(\frac{K_{1D}}{\sigma_{yd}} \right)^2$$

where σ_{yd} is the dynamic yield stress (as defined in the Nomenclature list), K_{ID} is the critical dynamic stress intensity, and B is the section thickness.

Using $\beta = 0.6$ in the above equation raises A (the temperature relative to NDT temperature) to a reasonable value for 1.0 in. thick sections. For example, with $\beta = 0.6$ and a 1.0 in. section thickness, the critical flaw size would be larger than 0.15 in. at yield stress loading.

Recommended procedures for selecting and qualifying Category II steels are discussed in Sec. 5.2.1. The basic requirements are modified for special situations in which reduced loading rates are assumed (Sec. 5.2.2), the sections used are thin (Sec. 5.2.3) or full-scale destructive testing is performed (Sec. 5.2.4).

The following relationship⁹ must be used if a K_{ID} curve is constructed from Charpy data:

where

 K_{1D} = dynamic fracture toughness in psi \sqrt{n} . E = modulus of the steel in psi C_V = Charpy V-notch measurement in ft-lb.

from Table II of Pathfinder
Report No. ACNP-62025,
(Page).

$$CV = 15.0 \text{H-Hz} (\text{Multimum value})$$

 $= 16.0 \text{H-Hz} (\text{Average value})$
 $E = 28 \times 10^6 \text{ psi} \text{ ASME Code}.$
 $KID = \sum \times 15.0 \times 28 \times 10^6 \text{ J/z}$
 $= 45.83 \times 10^3 \text{ psi} \text{ Jiu}$
 $= 45.83 \text{ Ksi} \text{ Jiu}$
 $Tyd = Tys + 30.0$
 $Tys = 40.5 \text{ Ksi} \text{ Table II Page}$
 $Tyd = 40.5 + 30.0$
 $= 70.5 \text{ Ksi}$
 $D = \frac{45.83}{70.5} = 0.65.$



BY 1. Husain DATE 916/89 SECTION 2.3.4-6 ОНКО 946 DATE 10/22/89 SUB-SECTION OF PROJECT NO 4-228 TASK NO PHISBA PAGE _____ OF ____ TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Container

1

TABLE 5. Category 11 fracture toughness requirements and criteria for ferritic steels with yield strength no greater than 100 ksi.

Required degree of safety (see Appendix C)	Large margin of safety.				
R squired amount of fracture toughness (nee Soc. 5.2)	Sufficient to prevent fracture initiation of cracks under dynamic loading.				
Thickness (8) (is.)	Criteris for meeting toughness requirements®				
0.625 to 4.0	* With full dynamic loading rates, NDT temperature ^b must be less than a maximum value. See Fig. 6 and Sec. 5.2.1.				
	. With reduced loading rates, NDT temperature can be determined from Fig. 7.				
	See Sec. 5.1.1.				
0.19 to 0.625	* Use DT Test E-606-80. 50% or greater shear fracture appearance required at LST. See Sec. 5.2.3.				
	Or				
	• Use DWTT Test E-436. 50% or greater shear fracture appearance required at LST. See Sec. 5.2.3.				
	Or				
	* Use any normalized steel made to "Fine Grain Practice" or better ^c . See Sec. 5.2.3.				
Loss than 0.19	* No requirements when B is less than 0.19 in. See Sec. 5.2.3.				

"Full scale destructive testing on a case-by-case basis may be used as an alternate to requirements listed below. See Sec. 5.2.4.

^bNDT is measured according to ASTM E-208, or as equivalent NDT can be established by subtracting 50°F from the midpoint of the 5/8 in. DT energy transition curve measured according to ASTM E-604 or the NDT temperature requirement can be met by selecting the maximum NDT temperature gives in Fig. 1 or Table 3.

"Steel with an NDT temperature lower than steels made to a fine grain practice.



Husani DATE 9/6/89 SECTION 2.3.4-DATE 10/22/89 SUB-SECTION ____ CHKD PROJECTINO NA.4-22 D TASK NO PHISPA PAGE TITLE NSP - PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shitting Containe

5.2.2 Qualifying Procedures for Reduced Loading Rates (0.625 in. to 4.0 in. thick sections)

Effective impact limiters have three major characteristics:

 They provide protection in all drop orientations.

They absorb all the kinetic energy from an impact.

 They dissipate kinetic energy at low force levels (on the order of 50 to 100 times the weight of the container).

For Category II, a temperature shift in the K_{1D}/σ_{yd} reference curve can be introduced if impact limiters are used to reduce the loading rate and protect fracture critical components. These energy absorbers can reduce the loading rate to well below that of the full dynamic level used to establish the K_{1D} curve. For loading rates that produce strain rates on the order of 10^{-1} in./in./s (typical for energy absorbing systems), the appropriate K_{10}/c_{yd} curve shown on Fig. 7 can be used instead of the K_{1D}/σ_{yd} curve to establish the maximum allowable NDT temperature. A temperature shift of 70°F can be used for low strength steels ($\sigma_{ys} < 60$ ksi); for higher strength steels (60 ksi $\leq \sigma_{ys} < 100$ ksi), a shift of 30°F can be used.



1. Husens DATE 9/6/89 SECTION 23.4.8 Ste DATE 10/22/85 SUB-SECTION ____ OF CHKD ____ PROJECT NO NO . 4 - 22 B TASK NO PH 1584 PAGE TITLE NSP - PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shitting Container.

From Figure 7, for Elloyd = 0.65



FIG. 7. Design chart for Category II fracture critical components.

For KI/SFYd = 0.65, A = - 20°(F) TNDT = 10.0°(F) Since TNDT = LST-A 10.0 = LST - (-20) LST = - 10°F <u>CONCLUSION: ALLOWABLE LOWEST SERVICE TEMPERATURE</u> <u>(Lowest metal Temperature) LST = - 10°F</u>



149111 DATE 916/89 SECTION 2.3.4 -9 DATE 10 2 185 SUB-SECTION ____ OF CHKD . PROJECTINO NO.4-22 B TASK NO PHISPH PAGE OF ... THE NSP - PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shitting Container .

For the outer vessel for HEXCEL Absorber containment, wall thickness = 3/8"

5 3.3 Qualifying Procedures for Thin Sections (up to 0.625 in. thick)

All sections between 0.19 in. and 0.625 in. thick have the same fracture toughness requirements. Any of the three following methods may be used to qualify Category 11 steels within this thickness range:

1. Application of full section thickness Dynamic Tear tests as specified in ASTM E-604-80. Steels will be qualified if 50% shear fracture appearance is achieved at the lowest service temperature.

2. Application of Battelle Drop Weight Tear Tests (ASTM E-436). If at least a 50% shear fracture appearance is observed at the lowest service temperature, then the fracture toughness of the steel is sufficient to qualify it.

 Use of any normalized steel made to "Fine Grain Practice" or better.

Brittle fracture is not considered a problem for sections less than 0.19 in. thick in Category II, and there are no fracture toughness requirements for these sections.



APPENDIX 2.10.6-B

Penetration Evaluation

OBJEC

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	DATE 913189 SECTION 216,10-1 CHO HT DATE 10/22/19 SUB-SECTION OF
	TITLE NSP - PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Contains
TIVE	- TO PERFORM PENETRATION ANALYSIS
ERIA	
tion and	alysis will assume a 13 pound steel cylinder

BC-TOP-9-1 Rev. 2

The penetration analysis will assume a 13 pound steel cylinder having hemispherical ends with a 1-1/4 inch diameter will be dropped from a height of 40 inches on various critical locations of the vessel (critical location of smaller thicknesses such as a noizle cap).

TIMETHOD OF APPROACH

vessel penetration calculation will be performed by hand calculations based on Ballistic Research Lab (BRL) and Stanford Research Lab (SRL) equations. These are described in Bechtel Topical Report 9A which has been reviewed and approved by NRC.

J HEFERENCE

0-8

BC-TOP-9A Revision 2 "Topical Report- Design of Structures for Missile (REF. 2.6) Impact; Bechtel Power Corporation, San Francisco, CA. September 1974 (REF. 2.6)

	Table	C-2	
PERFORATIO	N IN	STEEL	FORMER AS

Identification	Formula	Remarks	Equation No.
Eallistic Research Lab (Refs. 2, 3, 13)	$T^{3/2} = \frac{0.5 \text{ MV}_2^2}{17,-00 \text{ K}^2 \text{ D}^{3/2}}$		C-10
Stanford Research Institute (Ref. 20)	$\frac{E}{D} = \frac{S}{26,500} \left(26,000 \ T^2 + 2,500 \frac{V}{U_{\bullet}} \ T \right)$	See Limits page C-3	C-11

T . steel thickness to be just perforeted (in.)

M . mass of the missile (1b-sec2/ft),

V. . striking velocity of the missile normal to target surface (ft/sec),

K . constant depending on the grade of the steel, (K is usually . 1,)

D . diameter of the missile (in.)

E . critical kinesic energy required for perforation (ft-1b),

5 . ultimate tensile strength of the target minus the tensile stress in the steel (psi)

the second se

W . length of a square side between rigid supports (in.),

W. . length of a standard width (4 in.). (See Ref. 20)

BY 1Huspin DATE 913/89 SECTION 2. 0,10-2 011 -DATE 10 21 15 SUB-SECTION CHKD -PROJECT NO 4-226 TASK NO PHISBY PAGE TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Containe PENETRATION (Contd.)

The BRL Formula is shown below, modified by setting a material constant K = 1 and solving directly for steel plate thickness, T, which will just be perforated by the missile,



Weight of Missile = 13.0 lbs. Mass of $u = \frac{13.0}{32.17} = 0.4041$ lbs. Doop Height h = 40.0 in = 3.333 ft. Impact Velocity $V_s = \sqrt{2gh}$ $= \sqrt{2\times32.17\times3.333}$ = 14.645 ft/sec.

Diameter of Missile = D = 1.25 in

Steel Thickness to be just perfoonted

$$\frac{50.4041(14.645)^2}{2}^{2/3}$$
T = $\frac{672 \times 1.25}{672 \times 1.25}$

Available Thickness of Vessel Wall = 3.0 >> 0.015"

BY 1. HUSaux DATE 9/1189 SECTION 216.10-3 0+00 _____ DATE 10 22 (15 SUB-SECTION _____ OF _____ PROJECT NO 4-228 TASK NOPH 1584 PAGE _____ OF _____ THE NSP - PATHFINDER DECOMISSIONING STUDY Reactor Vessel Analysis - Type A Shipping Container. PENETRATION (Coutd.) ANALYSIS USING Stanford Research Institute Equestion External Kinetic Energy of Impact E = 2 MUSZ = 1 × 0.4041 (14,645)2 = 4.3.33 ft 163. = 0.52 IN.K S = 70,000.0 -0 = 70,000.0 pst. W= Conservatively Assume = 10:0 in $\frac{43.33}{1.25} = \frac{70,000.0}{46,500} \left(\frac{1600072}{1600072} + \frac{1500\times10.0}{47} \right)$ 0x T2 + 0.352BT - 0.0014 = 0 T= -0.3528 ± (10.3528)2+ 4x1x0.0014 ZXI = 0.0039 in 263.0" D.K 660.75" O.K CONCLUSION: As the versel wall is considerably greater thom calculated theckness of penetration, the effect of missile impact will be negligible on the versel wall and noggle Cover plate



APPENDIX 2.10.6-C

Load Resistance Calculations

DATE 9-14-29 SECTION 2.6.11-CATE 10 24 PS SUB-SECTION CHKD PROJECT NO. 4-228 TASK NO. PHISBY PAGE TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS I OBJECTIVE - TO PERFORM LOAD RESIS TANCE ANALYSIS I CRITERIA & METHOD OF APPROACH specifies a "Load Resistance" requirement IOCER Part 71 for Type Dpackages. Specifically required is to regard the package as a simple beam supported at its ends with a loading equal to five times its fully loaded weight. The allowable stress for this condition is the material yield strength. To be consistent with the ASME Code limits applied elsewhere in this report, the more restrictive ASME Code primary membrane stress limit will be applied. Flange E Er Saddle Saddle Max moment at B (Flange connection) MB = 2.676(51) + 5.389x41 + 7.04x31 + 11.2x21 + 12.38x1. + 13.73 x 0.1 = 96.63 K# = 1159.6 Kin, -Max Moment at C. Mc = 2.676 ×9.83 + 5.369×883 + 7.44×7.83+11.2×6.83+12:38×5.83 + 13.73×4.83 + 15.54×3.83+ 17.07×2.83+18.724×1.83 + 120.45×0.83+ 21.33×(233)2 = 507:35 KH = 6,08812 Km-Moment @F= 0.428x 5.1 + 1.096 x 411 3.37x 3.1+4.13 x2.1 + 4.28×1.1 + 4.281 (0.6)2 = 33.27 Kft = 375.3 Kin ~



BY 1. HUDOW DATE 9-14-89 SECTION 2.6.11-2 OHKD 187 DATE DIELES SUBSECTION OF _____ PROJECT NO. 4-22B TASK NO. FHISBA PAGE _____ OF _____ TITLE NSP - PATHFINDER VESSEL SHIPPINIC ANALYSIS Reactor Vessel Analysis as Type A' Shipping Counterly

Load Resistance (could.)

Max. Moment at E ME = 24.1×0.5 + 26:55×1.5 + 20.49×2.5+18:38×3.5

+ 9.89x4.5+ 6.51 × 5.5 + 4.29×6.5+ 428×7.5 + 4.13×8.5+ 3.37×9.5+ 1.1×10.5+0.43×11.25

= 391117 KH = 4,694,0 Km

May Moment et A = 9 911.68x 580.6 = 9,887.84 Kgt

9827.84= RC × 10.33 + RE X 24.0

957.2 = Re + 2.323RE ---- (1)

SP0:6 = Re+ RE ---- (2)

1. 323RE = 957.2 - 580.6 = 376.6

- RE= 284,6 K.

Rc = 580.6 - 294.6 = 296.0 K

max sheet is D@ x = 6.5 + 0.67 = 7.17' form C.

Maxo. Monneut at D = 2.68×17:0+5:39×16:0+7:44×15:0: (17:5'from A) 11:2×14:0+12:38×13:0+13:73×12:0+15:54×11:0

17.07x10.0+18.72x9.0+20.45x8.0+21.33x7.0

+ 20.9526.0+ 21.41× 5.0+ 22.49×40+24746×3.0

+24.46x2.0+24.46x.0 + 24.46(6)-12 2×12

- 296.0 % 7.17

MD = 100.86 KH = 1,210.4 Kin

1 Huran DATE 9-15-29 SECTION 2.1. 11-3 CHKD _____ DATE _ 10 22 59 SUB-SECTION _____ OF ____ PROJECT NO. 4 - 228 TASK NO. PHISBA PAGE . OF _____ TITLE NSP . PATHENDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Contains Load Revistance (costd.) Section Properties At section F I= I (53.754 - 47.754) = 154,5269144 section modulus $S_{xx} = \frac{154,526.9}{5375} \times 2$ = 5,749.8 in³ may moment for IGLOW = 375.3 Ki . Max Bending Strees 375.3 = D. 07 KSL Section modulus at C. DEF $I = \frac{1}{64} (1384 - 1324) = 2,849,992.4144$ Sxx = 2, 8 99, 992.4 x 2 = 42, 028.9, 13 map moment Mes Mosmo Mc = 6, 088.2 K.iu Map. Bending Stress = 6028.2 = 0.145 Ksi 50.07 . For 5 g Load Map Bending Stress in the Versel = 5 x0.145= 0.72 Ksi.

1. Hukan DATE 9/15/89 SECTION 2.6.11-4 DATE 10 22 PS SUB-SECTION CHKD . PROJECT NO 4-228 TASK NO THIS BA PAGE TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Contain At Flange Section 600" WELDING NECK FLAN BORT THRU . 45 HOLES For Bat Balts Min Rost Area A = .6.72 12 2 For 48 Bolts B = 48×6.72 = 322.56 · Equivalent Thickness for Dia. "BOAR THAUS. of 100 = t = 322.56 = 1.0314 Section Modulus Sxx = TT 82t = TT x50,02x1.03 = 8,064,0 in3 Mayo. Streps = 1159.6 = 0.144 K81 brig 8,064 for 19 " for 59 Load = 5x0. 144 = 0.72 Kst " " 3.528 " = 0.144x3 52 = 0.51 Kst Map shear Force 3 in the bott = 2.68+539 + 7.44 + 11.2 + 12.38 = 39.09K map. sheat stress in Batt for IG = 39-09 522.56 = 0.121 KH 4. to 5G = 5x0.121 = 0.61 KH 1 Ja 3.52 G= 352×0121 11 = 0.43K81
APPENDIX 2.10.7

References

Rev. 0

References

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- 2.10 "ASME Boiler and Pressure Vessel Code, Section III, Division 1, Subsection NB

LIST OF FIGURES

2.1 Manufacturers' Data Report Typical Hexcel Force Deformation Curve 2.2 2.3 Package Orientation - Cases 1 and 2 2.4 Hexcel Properties and Width and Depth of Crushing 2.5 Load Deformation Curve - Case 1 Side Drop Load Deformation Curve - case 2 Edge Drop 2.6 Typical ANSYS 3-Dimensional Finite Element Model 2.7 - Case 1 Side Drop Typical ANSYS 3-Dimensional Finite Element Model 2.8 - Case 2 Edge Drop Maximum Membrane Stress - Case 1 Side Drop - Crush Depth =1" 2.9 2.10 Maximum Membrane Stress - Case 1 Side Drop - Crush Depth =3" 2.11 Maximum Membrane Stress - Case 1 Side Drop - Crush Depth =5.11" 2.12 Maximum Membrane Plus Bending Stress - Case 1 Side Drop - Crush Depth =1" 2.13 Maximum Membrane Plus Bending Stress - Case 1 Side Drop - Crush Depth =3" 2.14 Maximum Membrane Plus Bending Stress - Case 1 Side Drop - Crush Depth =5.11" 2.15 Maximum Membrane Stress - Case 2 Edge Drop - Crush Depth =1" 2.16 Maximum Membrane Stress - Case 2 Edge Drop - Crush Depth =3" 2.17 Maximum Membrane Stress - Case 2 Edge Drop - Crush Depth =5.11" 2.18 Maximum Membrane Plus Bending Stress - Case 2 Edge Drop - Crush Depth =1" 2.19 Maximum Membrane Plus Bending Stress - Case 2 Edge Drop - Crush Depth =3" 2.20 Maximum Membrane Plus Bending Stress - Case 2 Edge Drop - Crush Depth =5.11"

FIGURE 2.1 MANUFACTURERS' DATA REPORT

FORM U-1A MANUFACTURERS' DATA REPORT FOR UNFIRED PRESSURE VESSELS
ALLE CALBORS ME. CO., TON ALLES, VLD. 2-4500-00000
Northers States POWEr MIDEL TETS, So, Daleta Pateriscor
TTHE WORKS TYPE SOLL CONSERVICE OF A LINE MALE AND THE DE NE 204 THE DEL
· SHELL. Math Math Mar Hat a the Tat a to the Date to the Date to the 100 - 100 - 10 man -
1. SEAMS Long Datt uft tol 4 255 " " Dongst. Lorented al of Course 2
HEADS BAMBOD HAPT SOLL . I A THE A CHART AND A CONTON OF THE CONCENTS
If removable, bolts used (baserial Same No. 7.5. Same Namer) (baserial Same Same) (baserial Same No. 7.5. Same Nomer) (baserial Same Same) (baserial Same Same Same Same Same Same Same Same
7. Constructed for (ER) persone of None None See Remarks
A NOZZLES
ATTE I Willie Bit Britter at Briter
10. INSPECTION Manholm, No Size Lessies
Threaded No Sim Location
11. SUPPORTS: SLIT - TE SE 1870"E & 1275"" (Desertes) (Part & Beer
12 REMARKS: BOILLE VETOF BUELVEF RELUCT
BO - 0.70" I.D. Gestrel Bod Open. Top Hoad - Veleod 6 - 5.70" I.D. LITLING Fipes Top Hoad - Selded
(Brief description of perpend of the vessel, or At Tenk, U Ate Tenk, L P.O. Be-basi Comment.) We certify that the matricestic made in this report and that all Create of antonial construction, and continuently of the real conton to just Abirt Conton to Uniform TERO 15 19 62 Signed All 10-Chalmore MSE. Co. By Some Co. Burger Date 10-Continent of Authorization Expires 10-Content of Content of Content of Content of Authorization Expires 10-Content of Authorization One 10-Con
Inspection Agency's Serial No.
VESSEL MADE BY
La Le Marine - Marine - Commissions - Bass o No. Be & Nortes

Copies of this Purd absainable from Amer. Bar Mast. Bar. 19 W. Doll D., Non York 18, N.Y.

FIGURE 2.2 TYPICAL HEXCEL FORCE DEFORMATION CURVE

The load and deflection characteristics encountered in statically crushing honeycomb are graphically shown in the figure below. This curve indicates the value of the resistance to loading offered by the honeycomb as the compressive load is overcome and crushing starts and continues. Three items on this curve are of special interest:

- A. COMPRESSIVE PEAK Honeycomb systems which are not precrushed or in which the initial contact area has not been reduced exhibit this peak force level at impact. This undesirable peak can be easily eliminated by proper design.
- B. MAXIMUM CRUSH LEVEL A horizontal line has been drawn across this curve representing a maximum crush level. If this hypothetical line had been developed as a design maximum, then the actual honeycomb fer value must remain equal to or below this level for crushing to take place.
- C. MINIMUM STOPPING DISTANCE The vertical line drawn down this graph represents a hypothetical stopping distance minimum established in conjunction with the maximum crush level value. This value represents a minimum value for the maximum crush level line and any crush level selected below the maximum limit will require an increase in this minimum thickness.



2-26





FIGURE 2.5 LOAD DEFORMATION CURVE CASE 1 - SIDE DROP (For Typical One foot Length of Package)





FIGURE 2.6 LOAD DEFORMATION CURVE CASE 2 - EDGE DROP (For Typical One foot Length of Package)



FIGURE 2.7 TYPICAL ANSYS 3-DIMENSIONAL FINITE ELEMENT MODEL CASE 1 - SIDE DROP

> ANSYS 4.4 SEP 22 1989 16:58:57 PREP7 ELEMENTS MAT NUM BC SYMBOLS

XV =1 YV =1 ZV =1 DIST=95.652 ZF =6 PRECISE HIDDEN



FIGURE 2.8 TYPICAL ANSYS 3-DIMENSIONAL FINITE ELEMENT MODEL CASE 2 - EDGE FROP



ANSYS 4.4 OCT 3 1989 14:03:21 PREP7 ELEMENTS MAT NUM BC SYMBOLS

XV =1 YV =1 ZV =1 DIST=105.496 XF =2.419 ZF =-34.5 PRECISE HIDDEN

PATHFINDER RPV EDGE DROP SHELL/GROUT MODEL - THETA = 0







2-35





PATHFINDER RPV SIDE DROP MODEL- DEPTH OF CRUSHING=3.0"





PATHFINDER RPV EDGE DROP MODEL DEPTH OF CRUSHING=1.0"



PATHFINDER RPV EDGE DROP MODEL- DEPTH OF CRUSHING=3.0"



Rev. 0

FIGURE 2.18 MAXIMUM MEMBRANE PLUS BENDING STRESS CASE 2 - EDGE DROP CRUSH DEPTH = 1 INCH



AN OCT 12: POS STEI ITEI SY CSYS DMX SMN SMX	6 1989 30:24 F1 STRESS P=2 R=1 (AVG) S=11 =0.349003 =-13445 =11453
XV	=1
YV	=-1
ZV	=1
DIST	(=105.496
XF	=2.419
ZF	=-34.5
A	=-12062
B	=-9295
C	=-6529
D	=-3762
E	=-996.019
F	=1770
G	=4537
H	=7303
I	=10070

PATHFINDER RPV EDGE DROP MODEL DEPTH OF CRUSHING=1.0"

FIGURE 2.19 MAXIMUM MEMBRANE PLUS BENDING STRESS CASE 2 - EDGE DROP CRUSH DEPTH = 3 INCHES



ANSYS 4.4 OCT 6 1989 12:50:20 POST1 STRESS STEP=7 ITER=1 SZ (AVG) S GLOBAL DMX =0.477837 SMN = -19272SMX =15287 XV =1 YV =-1 =1 ZV DIST=105.496 =2.419 XF =-34.5 ZF =-17352A =-13512 B = -9672C D =-5832= -1992E =1848=5688 =9528 H

=13367

I

PATHFINDER RPV EDGE DROP MODEL- DEPTH OF CRUSHING=3.0"

FIGURE 2.20 MAXIMUM MEMBRANE PLUS BENDING STRESS CASE 2 - EDGE DROP CRUSH DEPTH = 5.11 INCHES



ANSYS 4.4 OCT 6 1989 12:58:47 POST1 STRESS STEP=13 ITER=1 SZ (AVG) S GLOBAL DMX =0.483617 SMN = -22834SMX =17174 XV =1 =-1 YV =1 ZV DIST=105.496 =2.419 XF ZF =-34.5 =-20612 A B =-16160 =-11721 C D =-7275 =-2830 E =1615 F =6061 =10506 H =14952 I

PATHFINDER EDGE DROP MODEL- DEPTH OF CRUSHING=5.11"

LIST OF TABLES

- 2.1 Summary of Margins of Safety for Normal Conditions of Transport
- Major Vessel Components and Assemblies and Material Types 2.2
- Pressure Containing Parts and Support Structures Material 2.3 Properties
- Specified Minimum Mechanical Properties of Steel Materials 2.4
- 2.5 Summary of Design Stress Intensity Values, Sm
- 2.6 Hexcel Structural Properties
- 2.7 Weight, Velocity and Kinetic Energy Free Drop Analysis
 2.8 Load Deformation Characteristics
- 2.9 Maximum Stress Summary
- 2.10 Membrane and Membrane Plus Bending Margins of Safety

TABLE 2.1 SUMMARY OF MARGINS OF SAFETY FOR NORMAL CONDITIONS OF TRANSPORT

	Condition	Calculated Value	Allowable Value	Margin of Safety
2.6.1	Heat	Negliçible	Sm = 17.88 1.5Sm = 26.81	High
2.6.2	Cold - Max. Stress (ksi)	Negligible	Sm = 17.88 1.5Sm = 26.81	High
2.6.3	Reduced External Pressure - Max. Stress (ksi)	0.25	Sm = 17.88	49.4
2.6.4	Increased External Pressure - Min. Allow. Press. (psi)	20.0	437	20.9
2.6.5	Vibration	0.72	1.5Sm = 26.81	36.2
2.6.6	Water Spray Max. Stress (ksi)	Negligible	Sm = 17.88	High
2.6.7	Free Drop Max. Membrane Stress (ksi)	6.24	Sm = 17.88	1.86
	Max. Membrane Plus Bending Stress (ksi)	21.14	1.5Sm = 26.81	0.27
2.6.8	Corner Drop	N/A	N/A	N/A
2.6.9	Compression	N/A	N/A	N/A
2.6.10	Penetration Max. Thickness of Penetration (in)	0.015	3.0	199.0
2.6.11	Load Resistance Max. Stress (ksi)	0.72	1.5Sm = 26.81	36.2

NOTE: The margin of safety is defined as:

Margin of Safety = MS = (Allowable/Calculated) - 1

A positive margin of safety indicates that the allowable has not been exceeded.

TABLE 2.2 MAJOR VESSEL COMPONENTS AND ASSEMBLIES MATERIAL TYPES

			INDU							
Iten or					Manutac	nt Specs	Specs			
ABBORD IY	Tip. No	. Itens	Meteriel	MOT I.	Mig.	**10	Clad	Sonie	Part	Pen.
Top Flange Assembly	2	2	ASTR A 105-11 +55 Beld Over	1344		203x	152X 307X 365	14 7• X		173x
Upper Neck	2	3	ASTM A2 12-B -304L C 180	13 10	129*	1304	152x	132*x	16.5*	173•
Mein Cylinder	2	•	ASTM A2 12-8 +304L C 160	13 10	129X	130x 3 19x	152x	132x	16 5x	173x
Lower Head	2	5	ASTM A2 12-8 -304L C lad	13 1*	129*	130*	152X	132*x	165*	173•
BOTTON HEAD	2	Ð	ASTH A212-8 +304L C 160	13 10	129*			132*X		173•
Lover Cylinder	2	7	ASTH A2 12-8 + 304L C led	131•	129x	130x 319x	152x	(32*X		173x
Flue Ring	2	•	ASTH A 105-A +SS beld Over	134*		•	152X 307X	147•X		1738
Feedveter Nozz le Assy	2	12,26				207x 2 18x				133x
Feedbater Nozz le	2	12	ASTM & 105-11	134*			152X	147•X		1738
Thermal Sloeve	2	26	ASTM A 182 Type F 304L	217*				409 6-33*X		
Feedenter Pipe	2	67	ASTH A3 12 Type 3041 SS Seam less	193•				ADP 6-37*X		
Suction Nozz le	2	13	ASTH A 105-11 +55 Weld Over	134*			152x	147+5		173x
ischarge ozz le	2	14	ASTH A212-B +55 Weld Over	189*		203x	152x	132*x	16.5x	173x
nstrument iozz le	2	11	ASTM A 105-11 + Incone Over lay	134*			152X	147•x		173x
iquid Level	2	10	ASTH A 105-11 + Incone I Over lay	1340			152X	147•x		173x
team Dut let 012 le	2	9,62,63				213X 2./4X			16 5X	173x
ut let Nozz le	2	,	ASTH A 105-11 +SS Keld Over	134.			152x	147*X		173x
nterne l leeve	2	62	304L 55	150*		207x		132°x		173x



2-47

TABLE 2.2 (Continued) MAJOR VESSEL COMPONENTS AND ASSEMBLIES MATERIAL TYPES

Iten or					Hanutact	uring & P		nt Specs		
Assentiy	Fig. No.	Itema	Material	Met I.	Mtg.		Clad	Sonic	Port	Fluid Pen.
Thorne I S loove	2	63	ASTH A335 Gr P22 * SS Beld Over.	192*			2 10x	AOP 6-32*x		173x
Lower Head Assembly	2	5.6,7,8. 9.12.13. 14,26,67		129X 2 16X	130x 196x 203x 319x		152x		16.5x	173x
Upper Head Assembly	2	2,3			129x	130x	152x		16.5X	173x
Major Vessel Assembly	2	All Items			129x	3 19x 130x	152x		16 5x	173x
C losure Head	3	2	ASTM A2 12-8 - 3041 C 160	13 1+	129*			132*x		173x
Closure Flange	3	3	ASTM A 105-11 + SS Weld Over	134.			152× 307× 366×	147•x		173x
Control Rod Sleeve	3	4	ASTM E 167 (Incone Pipe)	169*				ACP 6-32*X		
Flanges	3	7,35	ASTM A 182 Type F 304	217.				147•x		
Liquid Level Nozzie	3	9	ASTM B166 (Inconel Bar)					ACP 6-31*X		
Instrument Nozz le	3	14	ASTM BIGG (Inconel Ser)					ADP 6-3 1*X		
Closure Assembly	3	All Items			129x	130x 319x 195x	152X		165X	173x 133x
tuds	•		ASTM A437 Gr 648	ACH 143*	2 16X	2 18X 260X		AOP 6-3 1*X	165x	
luts	.5		ASTM A437 GF CAT	ACH. 12*	2 16X 479X			ACP 6-3 1X	16.5X	
DISSUE Support	6		USS "TI"	322*	32 IX			132*x	16 5x	

TABLE .

NOTES:

· Indicates for material procurement or outside manufacture

X Indicates A-C shop procedure

All items were cleaned in accordance with A-C specification #43-201-661.

Identification of all items was maintained in accordance with A+C specification #43-101-282.

TABLE 2.3 PRESSURE CONTAINING PARTS AND SUPPORT STRUCTURES MATERIAL PROPERTIES

Component	1 ig.		Naterial		Strength (PSI)	Tensi le Strength LPSI)	Lion- petion	Charpy V-Notch Impact Emergy at 10 F (FT-LE)
Tep I lange	2	2	ASTH A 105-11	202	67,600 57,200	85,900 61,600	27.8 31.0	119 - 120 - 108 82 - 65 - 70
Upper Nock (4 peg.)	2	3	ASTN A2 12-8	200	42,100	71,100	м	17 - 16 - 15
Upher Meck (4 Seg.)	\$	8	ASTH A2 12-8	200	46 , 500	73,500 72,500	31	31 - 29 - 30
Main Cylinde #1	r 2	•	AST# A2 12-6	213	42,500	72,600	32	19 - 15 - 17
No in Cylinge	- 2	•	AST- A212-8	214	41,000	74,200 71,600	×	21 - 20 - 16
Deer Head	2	,	ASTH A2 12-0	201	47,000	71,800 73,000	30	26 - 30 - 31
is Ring	2	•	ASTH & 105-11	203	47,150	74.250 74,500	33.5 25	28 - 27 - 31 27 - 23 - 29
over y linder	2	7	ASTH A2 12-0	2 12	40,500	76.500	м	16 - 16 - 17
	3		ASTH A2 12-0	211	40,600	74,700	23	15 + 16 + 15
	2	•	ASTN & 105-11	209	45,000	73,000	<u>بر</u>	21 - 10 - 18
iquid Level	2	10	ASTN A 105-11	200	50, 100	74 ,000	32	16 - 20 - 19
netrument ozz le	2		ASTH A 105-11	210	\$1,500	74,000	25	18 - 17 - 15
	2	12	ASTH A 105-11	209	51,500	74,000	25	18 - 17 - 15
uction ozzie (1	2	13	ASTH A 105-11	205	42,500 40,500	72,500	M	17 - 18 - 22
uction 012 1 12	2		ASTH A 105-11	206	44,600 40,400	72,000	33	22 - 16 - 15
uction Dizia #3	2		ASTM & 105-11	207	43,700	71,750	35.5	18 - 19 - 19
ischnrpe ozzies 1,2,3	2	14	AST# A2 12-8	215 216 217	49,200	77,200 77,400 64,500	ж	23 - 21 - 24
ot Pad	2	25	ASTH A2 12-8	243	40,200	80,400	so	26 - 25 - 24
pport Luge	2	25	ASTH A2 12-0	244	45,100	76,200 77,500	\$1.5	16 - 16- 19
erma erve	2	26	ASTH A 182 Type F SOAL	246	29,100	77, 100	77	
.I Ring	2	27	ASTH A2 12-8	25)	\$1,500	79,600 76,400	22	20 - 20 - 16
Y	2	77	ASTH A2 12-0	234	49,100	80,900 76,400	دد	18 - 22 - 19
	2	62	ASTN 4240 Type 3041	231	32,950	76,927	64	





TABLE 2.3 (Continued) PRESSURE CONTAINING PA IS AND SUPPORT STRUCTURES MATERIAL PROPERTIES

.

				TABLE II				
Component	r 10-		Muterial	1	Tield Strangth (PSI)	Tons i to Strongth USI)	f ion- pation	Charpy Vinutch impact Limitary at 10 f (frick)
Fondenat ar Filpe	2	67	ASTH ASIZ Type SON Seemless	228	41,000	79,800	76	
C losure Head	3	2	ASTH A2 12-8	248	44,900	76,100 76,600	32	19 - 23 - 18
C losure F longe	3	3	ASTN A 105-11	247	46,700 45,300	70,000 71,500	40.5 38.5	110 - 140 - 126 130 - 115 - 80
Control And Sineve (1) pc Centrol And Sineve (9 pcs	3	3	ASTH 8167	249	36,000 33,500	89,000 91,500	45 48	
Flange (21) Flange (1)	3	;	ASTN A 182 Type F304	361 361	35,200 36,000	81,100 83,000	63 65	
Liquid Love! Norrie	3	•	ASTN 8 166	263	37,000	93,000	45	
Instrument Noss le	3	14	ASTN 8 166	262	42,000	96,500	*	
Instrument Flange	3	35	ASTN A 182 Type F304	362	35,750	81,350	67	
Stude Lot #1	•		45TH 4457 Gr 840	311-1	110,000	146 . 500 146 . 500	14	16 Ave (See Note 2)
Stude Lot 12	•		ASTN 4437 Gr 846	311-2	115,000	155,000 154,000	13	15 Ave (See Note 2)
NUTS LOT #1	5		ASTH 4437 Gr C40	m	103,750	125,000	20	32 - 28 - 30 (See Note 3)
Nuts Lot 12	5		ASTH 44.57 Br C40	968	101,250	126,500	20	35 - 34 - 32 (See Hote 3)
Support Columns 4" Top Fiste	•	2	"TI"	302	122,400	132,700 133,900	16 20	30 - 18 - 19 (See Note 4)
1-1/8" WEU	•	3	"TI"	209	119,400	127.800 125.600	20 21	32 - 33 - 35
1-1/4" Fienger		,	"T I"	300	106,000	1 16, 100 117, 100 (See Note 5)	19 18	30 - 30 - 31
	•	6,7,8	"T I"	Bartha a				
	6	18	" ¥ 1"	203	122,400	132,700 133,900	18 20	30 - 16 - 19 (See Hote 4)
1/2" Plates	•	20,22	ייו דיי	304	114,200 114,800	123,900	24 26	33 - 30 - 35
14" Top	•	21	*1 1*	305	106,630 106,200	117,600 117,800 (5- Hote 5)	24 23	м · м · м
II BOTTOM	•	25	יי די	301	106,000 106,000	116,100 117,100 (See Note 5)	19 10	30 - 30 - 31

HOTESI

1. Keyhole Impact at -50 F.

2. Charpy T-Notch Impects at room temperature.

3. The impacts at room temperature.

4. Approved deviation from specified 30 ft+Lb. Ave. Impact energy.

5. Approved deviation from specified tenalis strongth of 120,000 psi.

TABLE 2.4 SPECIFIED MINIMUM MECHANICAL PROPERTIES OF STEEL MATERIALS

Youngs Yield Tensile Modulus Material Strength Strength Specification Application E(103ksi) oy(ksi) ou(ksi) ASME SA-212, Shell and Hemispherical Grade B 28 38 70 Head

ASTM Tentative Specification (A212-61T)

TABLE 2.5 SUMMARY OF DESIGN STRESS INTENSITY VALUES, Sm

Stress Intensity, ksi (Multiply by 1000 to Obtain psi), For Metal Temperatures, deg F. Not to Exceed

Material	100	200	<u>300</u>	400	500	<u>600</u>	<u>650</u>	700
ASME SA-36	12.6	12.6	12.6		12.6	12.6	12.6	12.6
ASTM A-212 Grade B *	17.88	17.88	17.88	17.88	17.88	17.88	17.88	17.88

* Sm = Min. Actual Tensile Strength (Table 2.3)

= 71.5/4

= 17.88 ksi

TABLE 2.6 HEXCEL STRUCTURAL PROPERTIES

5052 Alloy Hexagonal Aluminum Honeycomb - Specification Grade

Hexcel Honeycomb Designation Cell — Material — Gage	Nominal Compressive						Crush Strength	Plate Shear					
	pct	pci Bare Stabilized psi	۳.	"L" Direction		"W" Direction							
	Cell — Material — Gage		Stren	ngih	Stre	ngth	Mod		Sue	ngth Si	Mod	Stre	ingth
3/8-50520025 1/4-5052004	3.7 7.9	typ 405 1420	min 270 970	1490	min 285 1050	typ 105 340	lyp 180 725	typ 260 700	min 200 650	typ 55.0 130	typ 170 440	min 115 390	typ 26.0 52.9

Notes: Test data obtained at 0.625 inch thickness.

TABLE 2.7 WEIGHT, VELOCITY AND KINETIC ENERGY FREE DROP ANALYSIS

Condition	Max. Tributary Weight	Velocity of Impact, ft/sec	Kinetic Energy of Impact			
Side Drop	24.5 k/ft	8.02	294.0 in.k/ft			
Edge Drop	291.1 k	8.02	3,493.2 in.k			

TABLE 2.8 LOAD DEFORMATION CHARACTERISTICS

Condition	Depth of Crush, in.	Percent of Available Crush Thkns.	Peak Deceleration Value, g	Maximum Resistance Load, k/ft
Side Drop	5.17	64.6	2.32	121
Edge Drop	5.11	63.9	2.35	1,457

TABLE 2.9 MAXIMUM STRESS SUMMARY

	Si	de Drop	Edge Drop			
Depth of Crushing, in	Max. Membrane Stress, ksi	Max. Membrane + Bending, ksi	Max. Membrane Stress, ksi	Max. Membrane + Bending, ksi		
0.5	2.56	5.33	3.03	13.06		
1.0	2.74	5.54	3.79	13.45		
1.5	3.35	6.77	4.63	16.44		
2.0	1.91	6.25	4.49	17.11		
2.5	2.13	6.97	5.01	19.10		
3.0	2.33	7.63	5.47	19.27		
3.5	2.52	8.23	5.90	20.78		
4.0	2.39	7.27	5.00	20.26		
4.5	2.54	7.77	5.30	21.47		
5.11	2.71	8.23	5.63	22.83		

2-53

TABLE 2.10 MEMBRANE AND MEMBRANE PLUS BENDING MARGINS OF SAFET?

Maximum membrane stress		5.90 ksi
Allowable membrane stress = Sm		17.88 ksi
Margin of safety = (17.88/5.90) - 1		= 2.03
Membrane plus bending stress	-	22.83 ksi
Allowable membrane plus bending = 1.5 Sm		26.81 ksi
Margin of safety = (26.81/22.83) - 1		= 0.17
Maximum tensile stress in closure head bolts	-	0.51 ksi
Allowable tensile stress in bolts = Sm		17.88 ksi
Margin of safety = (17.88/0.51) -1		= 34.06
Maximum shear stress in bolts		0.43 ksi
Allowable shear stress in the bolts = 0.6×38.0		22.80 ksi
Margin of safety = (22.8/0.43) -1		= 52.02

3.0 THERMAL EVALUATION

Since the package does not contain spent fuel, the release of fission products from a fuel source due to excessive temperatures does not need to be considered. Furthermore, since the decay heat gener-ation is conservatively calculated to be 4.67 watts, there are no significant internal thermal gradients that will lead to differential thermal expansion that will, in turn, result in stresses that must be considered in evaluating the structural integrity of the package. For a worst case analysis, a scenario was evaluated for the situation where the vessel and interior concrete are initially at a temperature of 100 deg F and the vessel instantaneously cools to -40 deg F (the maximum range of hot and cold environments specified in 10 CFR 71). In this case the vessel will shrink onto the interior concrete surface and stop further deformation. This is a self equilibrating phenomenon typical of thermal stress situa-For this worst case scenario, the calculated stress in the tions. vessel wall is 23.9 ksi which is less than the 1.5 Sm allowable for membrane plus bending stress of 26.81 ksi. The self equilibrating effect of this phenomenon, coupled with the thermal conductivity effects of the concrete heating the steel and relieving the stress condition is evidence that there will be no significant vessel damage.

The detailed calculations of the decay heat generation are shown in Appendix 3.1.1. The decay heat contribution from the other unlisted isotopes is negligible and was not shown.

The calculations of the thermal stress calculations are shown in Appendix 3.1.2.

Rev. 0

APPENDIX 3.1.1

Decay Heat Generation Calculations

Pathfinder Reactor Vessel and Internals Decay Heat Calculation

Isotope	Curies	Decay Heat (Mev/decay)	Energy Released (Mev/sec)	Released (Watts)
H-3	5.900E-01	5.683E-03	1.241E+08	1.987E-05
C-14	2.500E-01	4.947E-01	4.576E+09	7.331E-04
Fe-55	3.736E+01	5.700E-03	7.879E+09	1.262E-03
Co-60	2.567E+02	2.601E+00	2.470E+13	3.957E+00
Ni-59	1.490E+00	6.700E-03	3.694E+08	5.917E-05
N1-63	1.787E+02	6.700E-01	4.429E+12	7.096E-01
Nb-94	1.000E-02	1.719E+00	6.360E+08	1.019E-04
TC-99	1.000E-02	8.460E-02	3.130E+07	5.015E-06
Eu-152	1.300E-01	1.276E+00	6.138E+09	9.832E-04
Eu-154	1.000E-02	1.509E+00	5.583E+08	8.944E-05
	4.752E+02			4.670E+00

APPENDIX 3.1.2

Thermal Stress Calculations


BY HUBBLING DATE 10/16/89 SECTION 3.0-1 CHKD JYR ______ FATE 10/22/89SUB-SECTION _____ OF _____ PROJECT NO. 4-22B TASK NO. FHISBY PAGE ______ OF _____ TITLE NSP - PATHFINDER VESSEL SHIPPINIC ANALYSIS Reactor Vessel Analysis as Type A'ShippingContainer

- OBJECTIVE ' TO PERFORM STRESS ANALYSIS DUE TO THERMAL LOAD EFFECTS FOR NORMAL CONDITION OF TRANSPORT
- TI ASSUMPTIONS: ASSUMPTIONS ARE LISTED IN COMPUTATIONS WHERE THEY OCCUR

TREFERENCES - GENERAL REFERENCES ARE GIVEN ON PAGE 2-230F APPENDIXA SPECIFIC REFERENCES ARE PROVIDED WHERE EVER THEY OCCUR

NORMAL CONDITIONS OF TRANSPORT

Initial Conditions for all load cases-Initial conditions of ambient temperature at -20°F with no insolation and of ambient temperature at 100°F with maximum insolation

LOAD CASE 1 - Hot Environment #

The vessel will be structurally evaluated for an ambient temperature of 130°F in still dir and with maximum insolation



BY 1 HULAND DATE 10-16-39 SECTION 3.0-2 CHKD _____ DATE _____ SUB-SECTION _____ OF _____ PROJECT NO. 4 -- 2 2 B TASK NO. PHISB PAGE _____ OF _____ TITLE NSP . PATHFINDER VESSEL SHIPPING ANALYSIS Reactor Vessel Analysis as Type A' Shipping Contain LOAD CASE 2 - COLD Environment,* - The vessel will be structurally evaluated for cold amount temperature of - 40° F in still air and shade. As there is no internal heat generation and there are no constraints to thermal expansion stresses in the versel due to hot envisonment condition will be negligible. LOAD CASE-2-EVALUA TION In view of detail Thermal analysis and lack of ten perature profile, it will be conservatively assumed that curtially the vessel wall and internal growt gets heated to 100° = and than the temperature drops to -40°Fat which point the versal wall will try to contract and will be constrained to do so by the expanded concrete internals Conservatively Assuming Max Temperature Differential DT = 100-(-40) = 14007

EINEERINE COR				TIT	LE NS	P . 7	ATH	FIND	ER	VES	SEL	5	HIP	Chi2	G.	ANA
Thennal	A	na	lit	ist	10	out out	HJ	Sel	. Aw	alys	515 4	is T		e A'	shi	Priv
From	SUNT	-	Co	de		sec.	His	t.	T		Ab	per	J.	Į.	<u>-</u>	
	111					TAB	E 1-5.0	11.			+ ini	TR	et.	2.3	, ,	
				COE	FFICIE	NTS OF 1	HERMAN	Temper	ISION							
Materials	Coef- licient'	70	100	150	200	250 30	0 350	400	450	500	550	600 8.35	650	700	750 8 94	800 9.16
Carbon steel; Carbon moly steel; low-chrome steels (through 3 Cr)	e	6.07 0	6.13 .0023	6.25	6.38	6.49 6 .0140 .0	60 6.7 182 .022	6.82 .0270	6.92 .0316	7.02	7.12	7.23	7.33	7.44	7.54 .0616	7.65
Ferrous	Maturi	MOI		-32	STICIT	TA Y OF MA Modulus of 00 -10	BLE 1-6. TERIALS Elemienty, 0 70	5 FOR G E Value 200	Given X 300	EMPER 10' 199	ATUR	ES mperetu 600	(ire (F) e 700	Ref 	2.	3
Ferrous Carbon II C.30 or I	Materials eets with G	MOI		- 32 30.1	STICIT 5 -2 0 26	TA Y OF MA Modulus of 00 -10	BLE 1-6.1 TERIALS Electicity, 0 70 27.9	200 27.7	IVEN T Given × 300	EMPER 10' 155 400 27.0	ATUR	ES mperetu 600 25.7	(700 24.1	24		3
Ferrous Cerbon st 0.30 er H	Materials Reterials rets with C	MOI ai	DULI C	-32 30.1	STICIT 5 -2	TA Y OF MA Modulus of 00 -10	BLE 1-6.1 TERIALS Elemienty, 0 70 27.9	200 27.7	IVEN T Given × 300 27.4	EMPER 10' 155 400 27.0	ATUR a) tor To 500 26.4	ES mperetu 600 25.7	(ire (F) o 700 24.1	24		3
Ferrous I Carbon et 0.30 or H	Materials Anterials ents with C Ins. Dis Ni IN.		antent	-32 -32 30.1	STICIT 5 -2 0 29	TA Y OF MA Modulus of 00 -10 1.5 20.1	BLE 1-6.1 TERIALS Election, 0 70 0 27.9	27.7 27.7	1VEN T Given X 300 27.4	EMPER 10' 195 400 27.0	14 TUR	ES mperstu 600 26.7		10	2	3 •T
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Ferrous I Carbon II 0.30 or H MA	Materials Actorials Actorials Actorials Actor Ac		ontent xes = M	-32 30/	15 -2 0 20 2.11 =	TA Y OF MA Modulus of 00 -10 15 29 15 29 15 15 15 10 10 10 10 10 10 10 10 10 10 10 10 10	BLE 16.1 TERIALS Election, 0 70 27.9 27.9 2 V 2 V 2 V 2 V 2 V 2 V	200 27.7 27.7 27.7 27.7 27.7 27.7 27.7 2	10 EN T 300 274 274	EMPER 10' 155 400 27.0	26 4	ES mperstu 600 26.7) D		24	2	3 \$T
Ferrous I Cortoon at 0.30 pr to MA	Materials entry with c ix of when i human			-32 30.1 1-32 30.1 1-32 -32	in -2 20 20 20 20 20 20 20 20 20 20 20 20 20	TA Y OF MA Modulus of 00 -10 -1	BLE 16.1 TERIALS Elections, 0 70 27.8 2 27.8 2 2 27.8 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	200 27.7 27.7 27.7 27.7 27.7 27.7 27.7 2	1VEN T 300 274 274 1 T T 7.9	EMPER 10' 155 400 27.0	26 A	ES mporetu 600 26.7 D		10	28	3 .13
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4.0 CONTAINMENT

This chapter identifies the package containment for normal conditions of transport. Use of the term containment is predicated on the fact that the package source term consists of activated base metal and a thin internal corrosion film on the surfaces of the well confined internal structural components. In addition, the grout pumped into the vessel under pressure will also serve to bind the surface contamination within the interior of the reactor vessel. The package contains no free-standing pools of contaminated liquids, no explosive mixtures or potential aerosol particles that could be considered radiological hazard.

The exterior of the containment boundary is a steel structure which contains no valves or leak test ports. Leak test criteria or leak rate limits are not applicable to this package as a quantitative measure of containment integrity. Chapters 2.0 and 3.0 showed that there will be no rupture of the vessel due to normal conditions of transport from either structural or thermal stresses, and consequently there will be no release of activated or contaminated radioactive material.

4.1 Containment Boundary

The containment boundary consists of the reactor vessel, vessel head and the nozzle closure plates welded to the nozzle openings. The upper closure head contains 20 control rod drive nozzles, one liquid level and one instrumentation nozzle. The lower hemispherical head contains three pump suction nozzles and one feedwater nozzle. The lower cylinder assembly (part of the lower hemispherical head) contains three pump discharge nozzles and one steam outlet nozzle extending through the bottom of the cylinder assembly. There is a liquid level nozzle and an instrument nozzle connected to the shell of the vessel. Each of these nozzles will be seal welded closed with 3/4 in. thick plates.

4.1.1 Containment Vessel

The primary containment vessel is the reactor vessel itself and its closure head. The detailed description of the three inch thick vessel and closure head is presented in Section 2.1.1. The vessel and closure head were obviously designed to contain radioactivity as an operating power reactor vessel. The nozzles described in Section 4.1 will be seal welded closed to complete the containment boundary.

4.1.2 Containment Penetrations

There are no containment penetrations other than the seal welded closed nozzles as described above. There are no vents, valves or other openings in the containment boundary.

4.1.3 Seals and Welds

The containment does not contain any liquids, gases or materials that could leak out. Containment welds are described in Section 4.1.1.

4.1.4 Closure

The closure head is bolted closed with the original 48 closure studs and nuts designed to contain pressure and radioactivity during reactor power operation. The closure studs have been analyzed in Section 2.6.11 for stress loadings for normal conditions of transport, and shown not to exceed yield stresses for the one-foot drop case.

In addition, the Hexcel outer 1/4-inch thick shell is welded closed and cannot be inadvertently opened.

4.2 Requirements for Normal Conditions of Transport

The limit for radioactive material releases is defined in 10 CFR 71.51(a)(1) for normal conditions of transport. This regulation requires that there would be no loss or dispersal of radioactive contents, as demonstrated to a sensitivity of 10⁻⁶ A2 per hour, no significant increase in external radiation levels, and no substantial reduction in the effectiveness of the packaging.

4.2.1 Containment of Radioactive Material

The radioactive material contents are activated metal components and radionuclides in the thin corrosion product layer. The layer is formed on the internal surface of the reactor vessel and internals. A listing of the radioactive material contents is shown in Section 1.2.4. Activities are referenced to January, 1990. The total source term for the Pathfinder package is 476 Curies of activated material and 95 millicuries of surface contamination.

4.2.1.1 Activated Materials

Nearly the entire source term is comprised of activated carbon steel, stainless steel and Zircaloy-2 components that make up the Pathfinder package. The radionuclides are dispersed in matrices of component alloys which are stable under the conditions of normal transport. Therefore, the activated radionuclides will be contained within the metallic alloys under these conditions.

4.2.1.2 Corrosion Product Layer

The corrosion product layer is a thin, tenacious surface deposit which occurs on components that were in contact with the primary coolant. To evaluate the potential for release of this material, the 95 millicuries was divided by the allowable A2 value for the principal radionuclides of Co-60, Ni-63, Ni-59 and Fe-55 to compare the potential release to the 10⁻⁶ A2 limit. Accordingly, for the mixture of radionuclides:

```
F1+F2+F3+...Fn \leq 1

F Co-60 = 5.17 X 10<sup>-2</sup> Ci / ( 7 Co-60/A2)

= 7.39 X 10<sup>-3</sup> Ci / (100 Ni-63/A2)

= 4.23 X 10<sup>-2</sup> Ci / (100 Ni-63/A2)

= 4.23 X 10<sup>-4</sup> Ci / (900 Ni-59/A2)

= 1.44 X 10<sup>-7</sup> Ci / (900 Ni-59/A2)

= 9.45 X 10<sup>-4</sup> Ci / (1000 Fe-55/A2)

= 9.45 X 10<sup>-7</sup> Ci
```

Therefore,

A2 Quantity = 9.50 X 10^{-3} Ci / 7.81 X $10^{-3}/A2$ = 1.22 X 10^{-6} A2

The package is designed not to be breached. However, even in the most extreme case if all the surface contamination were instantaneously released from the package, the dispersal would be only slightly above the A2 limit. Obviously, this is a highly conservative case and actual dispersal would be much lower. As noted earlier, within the vessel, the interior of the vessel cavities will be filled with a grout to stabilize the pea gravel and to further bind the surface contamination to the vessel internals. The likelihood of all the surface contamination being released instantaneously is remote.

4.2.2 Pressurization of Containment Vessel

No gases or vapors are available to form an explosive gas in the reactor vessel. Therefore, containment pressurization is not possible. Containment internal pressure is evaluated in Chapter 2.0.

Rev. 0

4.2.3 Coolant Contamination

The Pathfinder package contains no coolant.

4.2.4 Coolant Loss

The Pathfinder package contains no coolant.

4.3 <u>Containment Requirements - Hypothetical Accident Conditions</u> Not applicable.

Rev. 0

5.0 SHIELDING EVALUATION

This chapter demonstrates that the shielding around the Pathfinder package meets the requirements set by 10 CFR 71 during normal conditions of transport. The controlling factor for shielding is to show that the exposure rate at two meters from the package will be less than 10 mRem/hr, and at the surface of the package will be less than 200 mRem/hr.

5.1 Shielding Design Features

5.1.1 Shield Description

The shielding used for the Pathfinder package consists of a 1-3/4 inch thick steel cylindrical shell welded to the vessel exterior shell, and a 1/4 inch thick steel shell welded around the exterior of the Hexcel impact limiter. The length of the 1-3/4 in. thick shield extends three feet below the bottom of the core elevation and five feet above the top of the core elevation. The overall length of the 1-3/4 in. shield is 14 feet. The extension above and below the core will shield the contribution from the core internals at the upper and lower regions of the core. There is a 1/8 inch thick shield covering the radial regions where the vessel support saddles and lifting sling points are located. The thinner shielding in these areas is to ensure impact loads will not be transmitted to the vessel shell. Figure 1.2 shows the general arrangement of the shielding materials around the reactor vessel.

In addition, the Pathfinder vessel was filled with 1/4 in. diameter pea gravel at the time of initial decommissioning. The pea gravel filled the vessel cavity with the apparent Apoption of one point at elevation 1289.5 ft on the northwest side ci he vessel. Based on measurements made at the side of the vessel there appears to be a void of approximately six inches diameter, with an exposure rate reading of 600 mR/hr. All other exposure rates are approximately 150 mR/hr at the same elevation. The pea gravel void fraction is assumed to be 25 per cent based on a correlation between calculated and measured exposure rates. In addition, no credit is taken for the grout fill material to be added prior to shipment for its shielding effectiveness other than filling in the apparent void with grout.

5.1.2 Shield Effectiveness

The combination steel shield and gravel materials are sufficient to reduce the calculated core axial centerline exposure rate at two meters from the package surface to approximately 6.9 mR/hr. The contact exposure rate (at one centimeter from the surface) is 24.9 mRem/hr. These exposure rates are well below the 10 CFR 71.47(c) limit of 10 mR/hr at two meters. Table 5.1 summarizes the maximum exposure rates.

Northern States Power Company made dose rate measurements at the outer surface of the vessel in the reactor cavity. The measurements were taken from the basement of the reactor building recording dose rath and elevations vertically upward. The measurements were made on the outer surface of the five-inch thick asbestos insulation. The results of the measurements show that the contact readings were all within the 10 CFR 71 limits of 200 mR/hr, ranging from a few mR/hr to about 80 mR/hr except for a band at the core beltline region that read between 100 to 200 mR/hr. One hot spot, had a reading of 600 mR/hr as noted earlier was located on the northwest side of the vessel at an elevation of 1289.5 ft. This hot spot appears to be a void in the gravel fill, and the grout fill is expected to shield the area sufficiently to meet the 10 CFR 71 limits. If grout does not fill the void, an external shield patch will be welded to the shield cylinder.

5.2 Source Specification

The primary radiation source present within the package is the gamma ray source resulting from the neutron activation of the metal components during operation of the reactor. The dominant radiation source is from Co-60 as confirmed by a field survey spectral analysis. The contribution of other activation radionuclides are negligible.

5.2.1 Gamma Source

The gamma ray source is dominated by the presence of Co-60 which is derived from the Co-59 impurity through an n-gamma reaction. Co-60 has two high energy gamma rays at 1.17 MeV and 1.33 MeV. The gamma ray energy spectrum that is present at the surface of the package will be the result of the attenuation of gamma rays emitted at these two source energies.

5.2.2 Neutron Source

The Pathfinder package does not have any neutron activity since there is no fissile material in the package. The spent antimonyberyllium source was disposed of in the reactor vessel, but the neutron activity is now negligible since the antimony-125 has a half-life of only 2.73 years and the reactor has been shut down since 1967.

5.2.3 Quantity of Radioactive Materials - Activation Analysis

An activation analysis was performed to determine the number of Curies of radioactive material in the vessel and internals. ORIGEN2 (Ref. 5.1) was the point activation analysis code used for this activation analyses. The Pathfinder plant's operating history, neutron flux, materials in the vessel and internals, and location of the components with respect to the core centerline were used to develop input for the ORIGEN2 code. Where specific information was not available from Pathfinder, references from other similar reactors were used with adjustments to account for differences in plant design or configuration. The ORIGEN2 code outputs provided the Curie per gram contents for each major region of the vessel and internals. These Curie per gram quantities were then multiplied by the number of grams in each component to obtain the number of Curies in each component. The total number of curies estimated is 476 Curies, 99 per cent of which is from Co-60, Ni-63, Fe-55 and Ni-59. The act: vation analysis was documented in a report entitled, "Radionuclide Inventory and Package Drse Rate for the Pathfinder Atomic Power Plant", prepared by TLG Engineering, Inc., October, 1989. A copy of this report is included in Appendix 5.5.1. The estimated radionuclide inventory is reproduced in Table 1.2, herein.

An estimate was also prepared of the surface contamination source term contribution to the gamma dose rate. Samples from the feedwater system piping to the reactor vessel were collected by Battelle Pacific Northwest Laboratories in 1980 and analyzed for surface contamination levels. The major constituents were Co-60, Ni-63, Fe-55 and Ni-59 as expected. Applying these measured interior surface contamination levels to the surface areas of each component of the vessel and internals showed the total inventory of surface contamination to be 95 millicuries. The detailed description of these calculations are also shown in the TLG Engineering, Inc. report in Appendix 5.5.1. The contribution of this internal surface contamination to the overall package dose is negligible.

5.3 Model Specification

5.3.1 Description of the Shielding Configuration

The concentric geometry of the Pathfinder vessel and internals lends itself to the use of cylindrical geometry for modeling. Vessel internals were represented by homogenized cylinders of the component materials and activation curies, radiating to its outer neighboring cylinder. In turn, each cylinder represents self-shielding to its interior cylinder neighbor reducing the contribution to dose reaching the vessel exterior.

In addition, the 66 cubic yards of pea gravel added at the time of initial decommissioning provides internal shielding. However, the density of the gravel is affected by the void fraction assuming uniform void fraction distribution within the vessel. The gravel void fraction is estimated to be 23.4 percent, and is conservatively assumed to be 25 percent for this analysis.

Rev. 0

An external steel shield was provided to reduce the external dose rates to meet 10 CFR 71 limits. The cylindrical shield extended from a point three feet below the bottom of the core to a point five feet above the top of the core. The extension above and below the core region was included to shield from the axial contribution at the top and bottom of the core region.

Two dose points were considered. The first was at the vessel outer wall core midplane region at a distance of one centimeter from the surface. This point was chosen to correlate the measured dose rates to calculated dose rates to confirm the calculational assumptic of distribution and gravel void fraction. The second dose point was also at the core midplane, but at a distance of two meters from the package outer surface. This latter dose point was selected to evaluate the dose limits in accordance with Title 10 CFR Part 71. Figure 5.1 shows the shielding geometry used for this analysis and the dose points selected for evaluation.

5.3.2 Shield Regional Densities

The shield design was based on a shield density of 7.8 gm/cm³ for iron. To keep the package weight to a minimum, the exterior shield was divided into two cylinders. The inner cylinder of 1-3/4 in. thickness welded to the vessel outer shell, and the outer shell of 1/4 in. thickness used as the Hexcel impact limiter container.

The shielding analysis was based on a 25% void fraction in the gravel grout mixture. Therefore, no credit was taken for the grout, except for shielding the hot spot.

5.4 Shielding Evaluation

The shielding evaluation was based on the ORIGEN2 activation analysis computer code results. These results were used as input to the Microshield (Ref. 5.2) computer code. All activation quantity results were decayed to January, 1990 to reflect the dose rates in the same year in which the vessel is expected to be shipped. Appendix 5.5 includes the report of the vessel shielding analysis and sample input for the Microshield code runs. .

APPENDIX 5.5.1

Radionuclide Inventory and Dose Rate Report



Prepared for

NORTHERN STATES POWER COMPANY

October 1989

Prepared by:

Francis W. Seymore PE

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Churdea Nor.us

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Page ii

TABLE OF CONTENTS

	PEVICICN100	Page
	EVECTTIVE CINAVADY	
	DI ID DOCE	IV
2	ADDDOVCH	
5	BATHENDER OPERATING HIERODY	
3.	MATERIAL COMPOSITION	
	MATERIAL COMPOSITION	
	4.1 Stainless Steel	
	4.2 Carbon Steel	
	4.3 Zircaloy-2	
э.	DISCUSSION OF THE POINT ACTIVATION ANALYSIS CALCU	LATIONS 8
	5.1 Activation Within the Core Boundary	
	5.2 Activation Radially Beyond the Core Boundary	
	5.3 Activation Axially Beyond the Core Boundary	
	5.4 Neutron Activation Model	
6.	COMPONENT DIMENSIONS AND MATERIALS	
	6.1 Superheater fuel element insulating tubes	
	6.2 Superheater baffle	
	6.3 Superheater support plate	15
	6.4 Superheater control rods	15
	6.5 Boiler fuel boxes	16
	6.6 Boiler shroud	16
	6.7 Boiler fuel element holddown structure	16
	6.8 Boiler CR guide tubes and remaining structure	16
	6.9 Boiler element poison chims	16
	6.10 Boiler control blades	10
	6.11 Instrumentation tubing and sample holders	10
	6.12 Roiler grid plate	
	6.13 Steam cenarators and support shelpes	
	6.14 Feedwater ring and support sherves	17
	6.15 Decrete verses ale dding	
	6.16 Reactor vessel cladding	
	6.10 Reactor vessel base metal	
-	6.17 Gravel	
1.	ACTIVATION ANALYSIS RESULTS	
	7.1 Specific curie contents	
	7.2 Curie contents by component	
8.	PRIMARY SYSTEM SURFACE CONTAMINATION	
	8.1 Methodology	
	8.2 Surface areas	
	8.3 Source terms	
1.18	8.4 Results	
9.	PATHFINDER VESSEL PACKAGE DOSE ESTIMATES	
	9.1 Nethod of Calculation	
	9.2 Results of Analysis	23
	9.3 Discussion	24
	9.4 Recommendations	25
10.	REFERENCES	20
	APPENDICES	
Ann	endix A ORIGEN2 Input and Output Description	20
Ann	endix B Pathfinder Ourie Estimate Spreadsheet	
Ann	endix C Microshield Input and Output Description	
ubbe	cheres conterestinete input and Output Description	



REVISION LOG



Page iv

EXECUTIVE SUMMARY

An activation analysis was performed for the Pathfinder Atomic Power Plant reactor vessel and internals in support of the Pathfinder Decommissioning Project. The results indicate that approximately 476 curies of neutron activation products, mainly Cooo and Ni63, are present within the reactor vessel and internals. The maximum curie density in any single reactor internal occurred in the boiler shroud, where the specific curie content was estimated to be 0.248 millicuries per gram.

The internal contamination ("crud") layer of the vessel and internals wetted surfaces contains approximately 95 millicuries, mainly Co60 and Ni63.

The external dose rate from the vessel package was calculated to be 6.8 mil per hour at two meters from the vessel at the reactor core midplane. This dose rate is based on 2 inches of carbon steel shielding on the vessel exterior, extending from five feet above the core upper elevation to three feet below the core lower elevation. This calculated dose rate also assumes an average packing density of the gravel present in the reactor vessel of 75% of the maximum.

The results of these analyses show that the Pathfinder reactor vessel and internals can be shipped intact as a single package without exceeding the LSA limits on specific curie content or US DOT external package dose rate.

Page 1 of 61

1. PURPOSE

This report details the calculations performed to support and provide data for the Pathfinder Atomic Power Plant decommissioning plan. They consist of a neutron activation analysis of the reactor vessel and internals, estimation of internal surfaces contamination levels, and estimation of dose rates from the package as currently envisioned for shipping.

The neutron activation calculations have been performed using the ORIGEN2 [1] computer code, which was obtained through Oak Ridge National Laboratory's Fadiation Shielding Information Center. The calculations determining the total curies based upon the ORIGEN2 specific curie values (curies per gram) were performed using the Lotus 1-2-3 version 2.01 [2].

The estimate of the primary system surface contamination was made using manual estimates of vessel and vessel internals surface areas, coupled with the Battelle estimates of specific surface contamination levels in the feedwater and main steam systems [3].

The calculation of the external exposure rates from the Pathfinder vessel package were performed using the Microshield code version 3.12 [4]. A variety of calculations were performed to bracket the possible amounts of gravel between the more highly activated portions of the reactor internals and the exterior of the package.

Page 2 of 61

2. APPROACH

The activation analysis has been performed using ORIGEN2. Input to the code included material compositions, the Fathfinder operating history and neutron fluxes. Wherever possible, data obtained through NSF [5,6,7] has been employed. The balance of the input data was extracted from the available literature and is referenced as such. Conservative assumptions were made and justified wherever actual plant data was not available.

The estimate of the primary system surface contamination has been made using manual estimates of vessel and vessel internals surface areas, coupled with the Battelle estimates of specific surface contamination levels in the feedwater and main steam systems [3].

For the shielding analysis, cylindrical source geometry was used as the input model to the Microshield code. Multiple concentric cylindrical regions were used to model the vessel and internals. The activation analysis results from ORIGEN2 were used as a source term for the vessel and internals.

Page 3 of 61

3. PATHFINDER OPERATING HISTORY

The Pathfinder Atomic Power Plant operating history has been extracted from data provided by Northern States Power. An available Pathfinder Six-Month Operating Report indicated that Pathfinder had its initial criticality on March 24, 1964. The plant's final shutdown occurred, due to a condenser tube failure, on September 16, 1967. The total core burnup was 16,635 megawatt(thermal)-days [5].

The Pathfinder Atomic Power Plant Safeguards Report [6] and subsequent correspondence [8] indicates the core design 100% power level to be 199.6 megawatts-thermal, at a core design average neutron flux of 3.5 E + 13 neutrons/cm²-second.

Since most of the power operation occurred towards the end of the plant's life, for the purposes of this analysis it was conservatively assumed that the plant operated at 100% power for 87.6 days, until September 16, 1967.

Page 4 of 61

4. MATERIAL COMPOSITION

Three material compositions were used in this analysis - Type 304 stainless steel, carbon steel and Circaloy-2. There are other materials used in the construction of the components in this analysis. However, from the research performed, there does not appear to be any component of significant mass manufactured from a material other than the three analyzed here, which would significantly affect the results of this study.

In some of the material composition tables included below it may be noticed that some rather conspicuous elements are missing. As an example, in 304 SS, carbon and oxygen are missing from the table. The source of some of this data [9] did not include certain elements because their activation would not contribute significantly to the long-lived isotopes of interest in a decommissioning study. Annotations have been made in the footnotes to the tables regarding the lack of certain isotopes, and their typical concentrations in the material have been provided.

4.1 Stainless Steel Type 304 [9]

With the exception of the boiler boxes, all reactor vessel internals and the reactor vessel cladding are assumed to be manufactured from Type 304 stainless steel. The composition of this steel for use as input to the ORIGEN2 activation code is listed below.

Element	ppm or %	Element	ppm or %	
Li	0.13	Y	5	
N	452	Zr	10	
Na	9.7	Nb	89	
Al	100	Mo	2.60%	
Cl	70	Ag	2	
K	3	Sb	12.3	
Ca	19	Cs	0.3	
Sc	0.03	Ba	500	
Ti	600	La	0.2	
v	456	Ce	371	
Cr	18.4%	Sm	0.1	
Mn	1.53%	Eu	0.02	
Fe	70.6%	Тъ	0.47	
Co	1414	Dy	1	
Ni	10.0%	Но	1	
Cu	3080	Yb	2	
Zn	457	Lu	0.8	
Ga	129	Hf	2	
As	194	W	186	
Se	35	Pb	67	
Br	2	Th	1	
Rb	10	U	2	
Sr	0.2			

Notes: No material certifications are available for the Pathfinder vessel internals; the elemental composition of 304 stainless steel above was taken from NUREG/CR-3474 [9]

Carbon is not included in this composition, but normally exists in Type 304 stainless steel at a maximum level of approximately 800 ppm. The creation of C^{14} from the $N^{14}(n,p)C^{14}$ reaction dominates the $C^{12}(n,\gamma)(n,\gamma)C^{14}$ reaction by several orders of magnitude.

The reactor vessel cladding may contain up to 0.8% niobium (columbium) [10]

Page 6 of 61

4.2 Carbon Steel [9]

The reactor vessel shell is manufactured from carbon steel. The composition of this steel for use as input to the ORIGEN2 activation code is listed below.

Element	ppm or %	Element	ppm or %	
Li	0.30	Y	20	
N	84	Zr	10	
Na	23	Nb	18.8	
Al	33	Mo	0.56%	
CI	40	Ag	2	
K	12	Sb	11	
Ca	14	Č\$	0.2	
Sc	0.26	Ba	273	
Ti	2	La	0.1	
V	80	Ce	1	
Cr	0.17%	Sm	0.017	
Mn	1.02%	Eu	0.031	
Fe	98.0%	Tb	0.45	
Co	122	Dy		
Ni	0.66%	Ho	0.8	
Cu	1274	Yb	1	
Zn	100	Lu	0.2	
Ga	80	Hf	0.21	
As	532	Ta	0.13	
Se	0.7	Ŵ	5.5	
Br	0.85	Pb	820	
Rb	48	Th	0.18	
Sr	0.15	Ű	0.20	

Notes: No material certifications are available for the Pathfinder vessel and internals; the elemental composition of carbon steel above was taken from NUREG/CR-3474 [9].

Carbon is not included in this composition, but normally exists in carbon steel at a maximum level of approximately 3000 ppm. The creation of C^{14} from the $N^{14}(n,p)C^{14}$ reaction dominates the $C^{12}(n,\gamma)(n,\gamma)C^{14}$ reaction by several orders of magnitude.

Page 7 of 61

4.3 Zircaloy-2 [11]

The boiler boxes surrounding the boiler fuel in the reactor are manufactured from Zircaloy-2. The composition of this metal for use as input to the ORIGEN2 activation code is listed below.

ppm or %	Element	ppm or %	
13	Fe	0.15%	
0.33	Co	10	
120	Ni	500	
80	Cu	20	
950	Zr	98.0%	
24	Nb	120	
35	Cd	0.25	
20	Sn	1.6%	
20	Hf	78	
0.1%	w	20	
20	U	0.2	
	ppm or % 13 0.33 120 80 950 24 35 20 20 0.1% 20	ppm or % Element 13 Fe 0.33 Co 120 Ni 86 Cu 950 Zr 24 Nb 35 Cd 20 Sn 20 Hf 0.1% W 20 U	ppm or % Element ppm or % 13 Fe 0.15% 0.33 Co 10 120 Ni 500 80 Cu 20 950 Zr 98.0% 24 Nb 120 35 Cd 0.25 20 Hf 78 0.1% W 20 20 U 0.2



Notes: The elemental composition of Zircaloy-2 above was taken from PNL-6046 [10].

5. DISCUSSION OF THE POINT ACTIVATION ANALYSIS CALCULATIONS

The point activation calculation of materials at the Pathfinder Atomic Power Plant was divided into two categories:

- Materials located within the radius of the core boundary.
- Materials located outside the radius of the core boundary.

For materials located within the core boundary, the ORIGEN2 BWR-U cross section library was chosen. This library contains one-group energy spectrum-averaged neutron cross sections, which best represent the neutron energy spectrum within the core. For materials located outside the core boundary, the ORIGEN2 thermal cross section library - which contains only thermal cross sections at 68 ° F. - was used.

A review of the information available on the thermal and fast neutron fluxes as a function of radial distance from the core boundary at Pathfinder indicated a high degree of spectrum thermalization beyond the core boundary, through and beyond the pressure vessel. Typically, more recent light water reactor (LWR) designs have pressure vessels two to three times the thickness of the Pathfinder vessel. While the large LWR's and Pathfinder both see significant thermalization in the plenum region just beyond the core boundary, the thick pressure vessel of the new, large LWR's significantly hardens the neutron flux energy spectrum. At Pathfinder, the neutron spectrum remains relatively well-thermalized beyond the outer wall of the pressure vessel.

Generally, the region inside the core boundary of typical light water reactors has an average fast/thermal neutron flux ratio on the order of 8:1. Outside the core boundary at Pathfinder, the fast/thermal neutron flux ratio was usually less than 1:5. Using in-core cross sections (the ORIGEN2 BWR-U library) would not be representative of the neutron spectrum, and would underestimate the activation of materials beyond the core boundary.

Use of the thermal cross section libraries (the ORIGEN2 thermal library) for radial regions beyond the core boundary requires several factors to be taken into account:

- The thermal cross section libraries contains cross section data for neutrons at room temperature (2200 m/second, or 68 * F). The curie content results calculated by ORIGEN2 must be temperature-corrected for regional temperatures outside the core boundary.
- Certain radionuclides in an activation analysis of reactor materials are products of fast neutron reactions only. There is only one radionuclide produced which has a significant impact on curie content and radiation levels - Mn⁵⁴. Manganese-54 is the product of the Fe⁵⁴(n,p)Mn⁵⁴ reaction, which a has significant cross section only above 1 MeV.

Since all of the radionuclides of interest in an activation analysis of reactor materials (with the exception of Mn^{54}) have temperature-corrected thermal cross sections which are greater than or nearly equal to the epithermal cross sections, the use of the thermal cross section library in conjunction with the total neutron flux will produce conservative results. In the case of Mn^{54}

the short half-life of the radioisotope (312.5 days) and the long decay time since shutdown (more than 20 years) imply that neglecting this isotope will not seriously impact the final results.

5.1 Activation Within the Core Boundary

Using ORIGEN2 and the BWR-U cross section library, point activation analyses were performed for Type 304 stainless steel and Zircaloy-2. A neutron flux of 3.50 E+13

a/cm²-second was used as input. The results were decayed to June 1, 1989 within the ORIGEN2 runs. Additional decay to January 1, 1990 was performed exterior to ORIGEN2 with Lotus 1-2-3 during the mass determination and total curie estimates.

5.2 Activation Radially Beyond the Core Boundary

The ORIGEN2 thermal cross section library was used in the point activation analyses of materials radially beyond the core boundary. Neutron fluxes for locations beyond the core boundary (at the core midplane) through the reactor vessel wall were taken from Reference 6. The fast and thermal fluxes from the reference figure were totaled, and the total used as input to ORIGEN2. The radial core midplane neutron fluxes are tabulated below.

Radial	Total flux
Location	(n/cm ² -sec)
Core boundary Boiler baffle Steam separators Reactor vessel inner wall	3.50 E + 13 1.16 E + 13 1.07 E + 12 3.30 E + 10
Reactor vessel outer wall	2.50 E+09

See Figure 5.1 for depiction of assumed axial flux profile.

For components located between the radial locations at which point activation analyses were performed, a logarithmic average of the ORIGEN2-calculated specific curie content results was used. Two such averages were computed and used. The first was at the location of the boiler shroud at a radius of 100.33 cm. The second was at the radial midpoint of the region from the core edge to the inner wall of the reactor vessel, an average radius of 127.6 cm. Point activation analyses were performed for Type 304 stainless steel and carbon steel. The results were decayed to June 1, 1989 internal to the ORIGEN2 code; additional decay to January 1, 1990 was performed with Lotus 1-2-3 during the mass determination and total curie estimates.

5.3 Activation Axially Beyond the Core Boundary

In the absence of axial flux maps for Pathfinder in regions above and below the core axial boundaries, flux reduction scaling factors were employed. For each 30 centimeter segment above and below the core, the total flux was conservatively reduced only by a factor of 10. In addition, for the regions axially above and below the core, point activation calculations were made with the thermal cross section libraries. For above and below the axial flux reduction factors to obtain corrected results.

See Figure 5.2 for depiction of assumed radial flux profile.

5.4 Neutron Activation Model

A model was developed for the radionuclide estimation of the Pathfinder vessel and internals; in general the features of the model matched the actual Pathfinder vessel and internals layout with the following exceptions:

- The reactor vessel lower head curvature was ignored; a straight, vertical wall was
 assumed to run the entire length of the model.
- Material more than 90 centimeters above the core or below the core was not included in the model.
- Neutron fluxes were identified for specific regions of the vessel and internals, i.e. the reactor core, the core boundary, the steam separator region, the vessel inner wall, and the vessel outer wall. Within each of these regions specific curie contents (curies per gram) were calculated using ORIGEN2 for each material present in the region. No effort was made to identify peaks or depressions in the local neutron flux levels and to incorporate these local flux disturbances into the activation mode.
- For the purpose of this estimate, the concrete bioshield was not considered in the activation analysis.

Figures 5.3 and 5.4 detail the neutron activation model used for the Pathfinder activation analysis.

Page 11 of 61

FIGURE 5.1

PATHFINDER ATOMIC POWER PLANT ASSUMED AXIAL FLUX



Page 12 of 61

FIGURE 5.2

PATHFINDER ATOMIC POWER PLANT ASSUMED RADIAL FLUX



Document N04-22-005 Page 13 of 61

FIGURE 5.3 PATHFINDER ATOMIC POWER PLANT NEUTRON ACTIVATION MODEL AXIAL PROFILE



Page 14 of 61

FIGURE 5.4

PATHFINDER ATOMIC POWER PLANT NEUTRON ACTIVATION MODEL RADIAL PROFILE



Page 15 of 61

6. COMPONENT DIMENSIONS AND MATERIALS

The Pathfinder reactor was a unique concept in early nuclear design, incorporating nuclear superheating of steam at the center of the reactor core. To achieve this superheating the reactor core was more complicated than current LWR designs, necessitating some unusual components, or components different from those in standard LWR designs. This section describes each component within the vessel package at Pathfinder. Unless stated otherwise, all components are Type 304 stainless steel.

It is important to remember that many of these components fall within two or more regions of the neutron activation model, and as such no single curie per gram value can be allocated to each component. Also, portions of the vessel internals which lay outside the neutron activation model were not considered in the model.

6.1 Superheater fuel element insulating tubes

Each fuel element in the superheater region of the reactor core was contained within an inner and outer insulating tube. The superheater fuel elements themselves were tube shaped, with a central flow path for the superheated steam. The insulating tubes surrounded the fuel element tube, providing additional flow paths for the steam to travel from the top of the reactor vessel through the superheater region and exiting from the vessel bottom. Water for neutron moderation surrounded the outside of the insulating tubes. The tubes run the entire length of the activation model. There are 468 sets of inner and outer superheater fuel element insulating tubes in the Pathfinder vessel internals. Fifty-two of the 468 fuel element/insulating tube sets also contained a superheater control rod.

6.2 Superheater baffle

The superheater baffle is a structural support for the superheater fuel elements as well as a flow baffle separating the water in the boiler section of the reactor core from the water surrounding the superheater fuel elements. It is 0.25 thick inch stainless steel and runs the entire length of the activation model.

6.3 Superheater support plate

The superheater support plate lies at the bottom of the superheater fuel elements, providing support and alignment of the elements. It is stainless steel and curved such that portions of the curved plate would lie outside the activation model; however all portions of the support plate were assumed to lie within the activation model.

6.4 Superheater control rods

The superheater control rods are solid stainless steel of approximately 0.79 inches diameter. Four clusters of 13 control rods are within the superheater fuel element insulating tubes.



Page 16 of 61

6.5 Boiler fuel boxes

Each boiler fuel element was individually contained within a Zircaloy-2 shroud to prevent cross-flow of water through the fuel assembly. The bottoms of the fuel boxes are manufactured from Type 304 stainless steel. There are 48 boiler fuel boxes, together containing 96 boiler fuel assemblies. There are 16 fuel boxes which contain 4 fuel assemblies apiece, and 32 fuel boxes which contain a single fuel assembly.

6.6 Boiler shroud

The boiler shroud is approximately 0.25 inch thick and surrounds the boiler fuel elements, separating the boiler reactor core section from the steam separator region of the vessel. It provides a flow baffle from the upflow through the boiler core region, and the down flow of the steam separator region.

6.7 Boiler fuel element holddown structure

The boiler fuel element holddown structure provides alignment of the boiler fuel assemblies, maintaining inter-assembly spacing for both fuel alignment and movement of the boiler control rods. The bottom of the holddown structure is an alignment grid to contain the fuel assemblies tops; the holddown structure also compresses the fuel assemblies lower nozzle springs, thereby providing both vertical and lateral alignment. The holddown structure also houses and provides alignment for the boiler control rod guide tubes.

6.8 Boiler CR guide tubes and remaining structure

This is basically the balance of the holddown structure above the lower fuel alignment grid; this section provides the rigidity necessary to compress the boiler fuel element lower nozzle springs and maintain the fuel alignment, both vertically and laterally. There are 16 boiler control rod guide tubes integral with the holddown structure.

6.9 Boiler element poison shims

Each boiler element boiler fuel box has indentations on each of it's four sides for the addition of a poison shim for use during initial core startup for control of excess fuel reactivity. It is estimated that 40 poison shims are present in the Pathfinder vessel at this time.

6.10 Boiler control blades

There are 16 cruciform-shaped control blades for the boiler fuel. Each blade is manufactured from borated steel; for the purpose of the activation analysis the boron present in the steel was ignored. The control blades entered from the top of the reactor core.

The six-month operation summaries for Pathfinder indicate several different rod bank heights for the boiler control blades were used during the core physics testing. For the activation analysis, a uniform rod bank height of 36 inches (one half out of core) was used. The 36 inches within the core was irradiated at full core flux, each 30 cm segment above the core was irradiated at 10% of the lower layers value, i.e. 10%, 1%, and 0.1% of full core flux.



6.11 Instrumentation tubing and sample holders

Three 0.75 inch outside diameter stainless steel instrument tubes run from the bottom to the top of the reactor core; in addition there are 12 small and 3 large sample holders at the elevation of the core. Without further information these items were combined into one item and assumed irradiated at the full core flux.

6.12 Boiler grid plate

The boiler grid plate provides lateral alignment for the boiler fuel elements as well as providing support for the fuel elements. The grid plate consists of two separate plates connected by bored-out pipes which surround the fuel element lower nozz es.

6.13 Steam separators and support shelves

The steam separators lie outside the core region at the same elevation as the core. There are 44 separators, grouped in sets of three; these are individually supported by three steam separator support shelves attached to the reactor vessel vall. Suction at the bottom of the reactor vessel draws water through the steam separator lower nozzle, which in turn draws water horizontally through the inlet nozzle in the side of the steam separator. The swirling action of the water separates the entrained steam which exits out the top of the separator.

Portions of the steam separator nozzles failed towards the end of operation of the Pathfinder reactor; these pieces were collected for examination, then returned to the vessel prior to final closure at the end of life.

6.14 Feedwater ring and supports

The feedwater ring is located at the bottom of the reactor vessel, beneath the steam separator support shelves. It is a ring header, oval in cross-section, which is perforated with holes to allow feedwater to be introduced into the lower plenum of the vessel. There the water mixes with water drawn through the steam separators prior to removal by the recirculation pumps and re-injection into the boiler section of the reactor core.

6.15 Reactor vessel cladding

The one guarter inch thick vessel cladding is made of SA 240 Type 304L stainless steel. As in the vessel shell, the cladding runs the entire vertical length of the activation model.

6.16 Reactor vessel base metal

The three inch thick outer wall of the reactor vessel is manufactured of SA212 Grade B carbon steel; it runs the entire vertical length of the activation model. The lower head of the vessel actually would lie within the model; but for simplicity the curvature of the lower head has been ignored, and the walls assumed to continue as a cylindrical shell to the bottom of the model.





Page 18 of 61

6.17 Gravel

The plant records indicate that approximately 66 cubic yards of pea gravel, weighing 214,000 pounds, were placed in the vessel after unloading of the fuel took place; this was done at the recommendation of the Atomic Energy Commission. The estimated void in the vessel after subtracting the volume occupied by the internals is approximately 75.6 cubic yards. This indicates an average packing fraction of approximately 87.3%. There will no doubt be local variations on this packing fraction. In addition, the void of 75.6 cubic yards does not consider the volume of piping beneath the vessel, and some of the gravel probably went through the core region and fell into these pipes.

7. ACTIVATION ANALYSIS RESULTS

7.1 Specific curie contents

The ORIGEN2 outputs provided the following curie per gram contents for the following regions and materials:

Region	Material	Ci/gram
Core (full flux)	304 SS	2.01E-4
Core (full flux)	Zirc-2	1.74E-6
Core (thermal flux)	304 SS	2.94E-5
Boiler Shroud	304 SS	2.48E-4
Steam Separators	304 SS	5.66E-6
Vessel cladding	304 SS	1.09E-6
Vessel wall (average)	SA212 CS	7.54E-8

 The increase in the curie per gram value for the boiler shroud relative to the in-core values is the result of the conservative assumption that the neutron flux is totally thermalized at this point.

The specific curie content of any individual portion of any component, except the boiler core shroud, within the activation analysis model meets the current definition of LSA waste, i.e. less than 0.3 millicuries per gram. The boiler core shroud, due to the conservative assumption of full thermalization of the flux, exceeds 0.3 millicuries per gram for the section of the shroud immediately adjacent to the reactor core. If the upper and lower sections of the shroud, which are irradiated at 10% of the flux level of the core region, are weight averaged with the central portion, the average millicurie per gram decreases to 0.248, which then meets the definition of LSA waste.

7.2 Curie contents by component

The results of the neutron activation analysis is presented in Table 7.1, and shows the radionuclides by component.

The reactor vessel cladding may contain up to 0.8% niobium [10]. At this level, the amount of Nb⁹⁴ present in the vessel cladding will be approximately 190 μ Ci. At a weight of approximately 4,240 pounds of stainless steel cladding in the activation analysis model,

this equals a concentration of 0.00078 µCi Nb94 per cubic centimeter. The limit for Class

A waste is $0.02 \,\mu$ Ci Nb⁹⁴ per cubic centimeter; therefore the concentration in the vessel cladding in the activation analysis model region alone is only 3.9% of the limit and will not present a problem for disposal.

TABLE 7.1

ESTIMATED RADIONUCLIDE INVENTORY IN THE REACTOR VESSEL AND INTERNALS AS OF JANUARY 1990

Component	#3	C14	1.55	Cath								Total
						#163	ND94	TCYY	EU152	Eu154	Others	Curies
Superheater baffle	0.08	0.03	4.58	34.40	0.19	21.97			0.02			
Superheater fuel insul. tubes	0.15	0.06	8.77	65.92	0.35	42.11			0.05			01.3
Superheater support plate									0.05			117.6
Superheater control rods			0.04	0.32		0.21						<0.01
Boiler fuel boxes	0.01	0.03	0.84	7 10	0.04	1 75						0.6
Boiler shroud	0.15	0.06	0.50	51 61	0.30	44.84					0.86	13.3
Boiler hold down structure	0.01		0.61	4 50	0.00	40.04				•••	0.01	108.6
Boiler CR tubes/remain, struct.			0.01	1.20	0.02	2.93		•••		•••	•••	8.1
Boiler element poison shime	0.03	0.01	1.00	3.09	0.02	2.30			•••	•••	0.01	6.6
Bailer control binder	0.05	0.01	1.00	14.14	0.08	9.03		•••	0.01	•••	•••	25.2
borter control blades	0.11	0.04	6.39	48.03	0.26	30.68	•••		0.03	•••		85.5
Instrumentation/sample holders		•••	0.48	3.58	0.02	2.29					0.01	6.4
Boiler grid plate	0.04	0.02	2.52	18.96	0.10	12.11			0.01			33.8
separators & supports		•••	0.54	2.93	0.02	2.66					0.01	6.2
ater ring & supports	•••	•••	0.15	0.83		0.75					0.01	1.7
vessel cladding			0.06	0.34		0.31						0.7
Vessel			0.41	0.14		0.09						0.1
												0.0
Total by isotope (curies)	0.59	0.25	37.36	256.66	1.49	178.68	<0.01	<0.01	0.13	<0.01	0.93	476.1
Percent of total by isotope	0.12%	0.05%	7.85%	53.91%	0.31%	37.53%	<0.01%	<0.01%	0.03%	<0.01%	0.19%	
Page 21 of 61

8. PRIMARY SYSTEM SURFACE CONTAMINATION

8.1 Methodology

An estimate was prepared to determine the total amount of radioactivity present in the surface contamination ("crud") layer on the wetted area of the reactor vessel interior and internals. A surface area for the vessel interior was calculated using dimensioned drawings. The surface area for the vessel internals proved impossible to accurately determine with the available drawings. An assumption was made as to the average thickness of the internals structural components; using this, and with the calculated and/or stated weight of each internals component, the total surface area can be estimated.

8.2 Surface areas

Calculations were performed to determine the wetted internal surface area of the reactor vessel and internals. The total estimated interior surface area of the reactor vessel is approximately 660 square feet. Using the weights stated on the drawings provided by NSP for all internal components, an effective surface area for the internals was calculated. For an average thickness of one-quarter inch, the 44,600 pounds of the internals equates to approximately 8,660 square feet. The total of both areas, converted to square centimeters, yields an affected surface area of 8.67E + 6 square centimeters.

8.3 Source terms

In 1980, Battelle Pacific Northwest Laboratories took samples throughout the Pathfinder Plant. There were three samples taken in the feedwater system; these were used as being representative of the type of interior contamination within the reactor vessel [3]. PNL data, when decayed to January 1, 1990, yields the following surface contamination levels:

Ladionuclide	pCi/cm ²
Na22	< 0.0077
Fe55	109
Ni59	15
Co60	5965
Ni63	4882
Nb94	< 0.25
Ag108m	< 0.09
Sb125	< 0.01
Sn126	< 0.06
Cs134	< 0.002
Cs137	< 0.06
Eu152	< 0.2
Eu154	< 0.45
Eu155	< 0.04
Ho166m	< 9.55
Ra228	< 0.3
Pu238	< 0.00093
Pu239	< 0.0039

8.4 Results

Using the total surface areas for the wetted portion of the reactor vessel and internals, and the areal curie density of each radionuclide measured by Battelle, total curies for each radionuclide was determined.

Radionuclide	Total curies	% of Total
Fe55	9.45E-4	1.0
Ni59	1.30E-4	0.14
C000 Ni63	5.17E-2	54.4
14105	4.436-4	44.5

All radionuclides with a total activity of less than 0.1 millicurie (i.e. < 1.0E-4 curies) have been ignored. The total estimate of surface contamination radionuclides within the Path-finder reactor vessel is 95 millicuries.

Page 23 of 61

9. PATHFINDER VESSEL PACKAGE DOSE ESTIMATES

9.1 Method of Calculation

Using the total curies per component estimated with the Lotus 1-2-3 spreadsheet and the geometry of the internals, the dose rate was estimated for a point external to the vessel. Given the multiple layers of sources represented by the concentric nature of the internals arrangement in the Pathfinder vessel, many different runs were necessary, since each source is also a shield for other sources in the internals. All dose rate models used cylindrical sources with cylindrical shields.

In addition to the self-shielding and shielding provided by the internals components and vessel, the gravel added at shutdown serves as an effective shield. However, the density of the gravel undoubtedly varies throughout the region of the neutron activation model. With 66 cubic yards placed into a free void of 75.6 cubic yards, the average gravel packing factor is 87.3% of the theoretical maximum. The maximum density is estimated to be 120 pounds per cubic foot (1.92 grams per cubic centimeter), based upon 214,000 pounds weight of the 66 cubic yards. Since this is similar to the density and composition of concrete, the shielding parameters for concrete were used in the Microshield problems.

Two dose points were considered. The first was at 1 centimeter distance from the vessel outer wall at the core midplane. This determines the maximum contact dose rate and can be readily compared to the measured doses that NSP collected in May of 1989. The second dose point is on the same plane as the first, the core midplane, but is two meters from the edge of the shielded package. The package in this case includes the impact absorbing material and carbon steel shield which will surround the vessel for shipment. The impact absorbing material, together with any radiation shielding, will add approximately ten inches to the radius of the vessel package. This second dose point corresponds to the measuring point necessary to meet the U.S. Department of Transportation (US DOT) regulations.

In addition to the direct dose estimate from the core height region of the vessel internals and vessel wall, the regions above and below the core region are also neutron activated and must be considered in the dose rate. Calculations were performed which indicated, in the most conservative case (no shielding), that the total dose rate at two meters from the package side increased approximately 15% above the dose calculated for the core region. Rather than calculating a dose rate for each shield thickness for the regions above and below the core, the dose rate from the core midsection was increased by 15% to account for the contributions of the regions above and below the core.

See Table 9.1 for measured dose rates on the side of the Pathfinder vessel as of May 1989.

9.2 Results of Analysis

The unshielded dose rate at the core midplane on contact with reactor vessel, assuming a 75% gravel packing factor, is estimated by the activation analysis model to be approximately 254 mR per hour.

With the equivalent of two inches of steel shielding, and assuming a gravel packing factor of 75%, the external radiation exposure rate at the core midplane 2 meters from the exterior of the impact absorbing shell will be 6.8 mR per hour. The dose rate at 1 centimeter from the exterior of the impact absorbing shell (i.e. contact dose rate) at the core midplane will be 24.9 mR per hour. These dose estimates meet the US DOT requirements for external dose rates.

See Figure 9.1 for a depiction of vessel package radiation exposure rates versus the percentage of gravel in the interior void spaces at 1 centimeter from the vessel outer wall. Figure 9.2 shows a depiction of the vessel package radiation exposure rates versus the percentage of gravel in the interior void spaces for three shield thicknesses, as well as unshielded, at a distance of 2 meters from the package.

9.3 Discussion

Given the uncertainties present as to the actual configuration of the gravel within the vessel interior, the best that can be stated about the results shown in section 9.2 is that the measured dose rates alongside the vessel exterior are reasonably close to those estimated with the assumption of 75% gravel packing factor (without external shielding). Since the lowest core midplane dose rate is on the east side of the vessel package and is approximately 40% less than the dose rate estimate of 254 mR per hour, one approach might be to use this as a bench mark and adjust all dose rates accordingly. This would compensate for the built-in conservatism in the activation analysis model. For the purpose of the shield design, this was not done, the recommended shielding thickness is based solely upon the ORIGEN2 activation analysis and Microshield dose model.

The shield thickness of two inches obviously runs the full length of the core region. The shield should also run five feet above and three feet below the core region as well. The five feet above will encompass the 100 mR per hour peak on the west side of the vessel, which occurs approximately 2 feet above the top of the core. With the five additional feet of shielding above the core, all dose rates measured on the side of the vessel above this point meet US DOT limits. For the section below the core, the three foot length is based more upon mechanical considerations than dose limits, as the support feet for the vessel can serve as attachment points for the lower portion of the radiation shield. Based upon the dose readings in Table 9.1, a two foot length would be sufficient to meet US DOT radiation limits; but the three foot length will be easier to implement.

The hot spot shown on the northwest section of the vessel, as listed in Table 9.1 is probably a region of gravel with a packing factor lower than average. If this is the case, the grouting operation will almost certainly close this void region and lower the dose rates. Since the grouting will be performed prior to vessel lift, this will be determined prior to the loss of shielding around the vessel now presently provided by the bioshield. The grouting of the vessel will reduce the dose rate to that comparable for the remaining sections of the reactor vessel package at the same elevation.

The decommissioning crew should be prepared for the possibility that the dose rates will be greater than those estimated to be present with the recommended two inches of steel shielding. If this occurs it is likely to be on a limited basis, most probably one or two small areas. Additional steel shielding will then have to be used to further reduce the radiation levels to meet 10 mR/hour at 2 meters from the package.

5.4 Recommendations

The project should plan for the fabrication and use of a two inch thick steel shield surrounding the reactor vessel, starting at three feet below the bottom of the core, and ending five feet above the top of the core. The two inch thickness can include any additional steel to be placed around these regions, such as the impact absorbing material.

The vessel interior should be grouted to reduce the possibility of voids or "windows," which would allow excessive radiation leakage.

If the opportunity occurs, additional chara sterization would be invaluable in reducing the uncertainties with this hot-spot region on the northwest side of the vessel. Detailed geometry studies would help assess the extent of void assumed to be present.

Page 26 of 61

TABLE 9.1

PATHFINDER MEASURED RADIATION LEVELS AS OF MAY 1989

Elevation (MSL)	South mR/hr	West mR/hr	North mR/hr	East mR/hr
1285	1.75	2	5	1.8
1286	3	8	5	2.2
1287	9	27	20	7
1288	33	80	225	30
1289	90	175	450	150
1290	45	75	140	80
1291	17	85	35	22
1292	15	90	18	15
1293	9	95	10	10
1294	6	100	ž	
1295	3	75	3.8	32
1296	1.8	35	2.8	2
1297	1.5	8	2.2	15
1298	1.7	2.2	15	16
1299	1.7	1.2	15	1.0
1300	5			
1301	3		5	
1302	3	15	25	25
1303		3	3	18
1304		ž		1.0
1305		'2		

The highest observed dose rate was 600 mR/hour at elevation 1289.5 elevation, on the northwest side of the vessel.

Page 27 of 61

FIGURE 9.1

GRAVEL PACKING & VS EXPOSURE RATE CONTACT DOSE



Dose rate is measured at a point 1 centimeter from the vessel outer surface directly adjacent with the reactor core midplane.

Page 28 of 61

FIGURE 9.2

PATHFINDER VESSEL PACKAGE GRAVEL PACKING % vs EXPOSURE RATE DOSE AT 2 METERS



steel shield	-0- 2.5 in steel shield

Shield			Gravel Packi	ng Factor, %		
Inickness	0	25	50	75	90	100
Dose rate (mR/hr)						
0.0 inch	498.2	120.2	65.49	45.98	39.08	35.56
1.5 inch	128.1	29.04	15.73	11.03	9.377	8.54
2.0 inch	80.18	18.04	9.768	6.845	5.825	5.303
2.5 inch	49.97	11.16	6.041	4.234	3.603	3.282

Dose rate is measured at a point 2 meters from the vessel package directly adjacent with the reactor core midplane. Values include an additional 15% for the portions of the activated reactor core above and below the active fuel section.

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- ACNP-5905, "Pathfinder Atomic Power Plant Safeguards Report, Part II License Application." Northern States Power Company, October 1960.
- 7. Op. cit., ACNP-5905, Figure 4.4.
- Letter NSLTG-89-110, Kuroyama to LaGuardia dated September 1, 1989, with attachment Page iii of ACNP-5905, "Pathfinder Atomic Power Plant - Safeguards Report, Part II License Application." Northern States Power Company, January 1962.
- NUREG/CR-3474, "Long-Lived Activation Products in Reactor Materials", Evans, J.C., Pacific Northwest Laboratory, August 1984.
- ACNP-62025, "Pathfinder Atomic Power Plant Reactor Vessel Materials, Fabrication and Inspection", Patterson, J.F. and Potochnik, J.F., Page A-147, Allis-Chalmers Manufacturing Company, December 1, 1962.
- PNL-6046, "Spent Fuel Disassembly Hardware and Other Non-Fuel Bearing Components: Characterization, Disposal Cost Estimates, and Proposed Repository Acceptance Requirements", Luksic, A.T., et. al., Pacific Northwest Laboratory, October 1986.



Page 30 of 61

APPENDIX A

ORIGEN2 INPUT AND OUTPUT DESCRIPTION

Page 31 of 61

APPENDIX A

ORIGEN2 INPUT AND OUTPUT DESCRIPTION

The calculation of neutron-induced activity in the Pathfinder pressure vessel and internals was performed using a PC version of the point neutron activation and depletion code ORIGEN2. This version of the code was originally obtained from the Oak Ridge National Laboratory (ORNL) as ORIGEN2-PC Version 1.00, and recompiled under Microsoft's FORTRAN 4.01 optimizing compiler. After recompilation, the code was verified using the sample problem given in the ORNL-provided manual.

Some of the problem-specific data required as input for ORIGEN2 include the following:

- 1. Material compositions.
- 2. Neutron flux exposure histogram.
- 3. Total neutron flux (or specific power level).
- 4. One-group neutron cross-section libraries, decay libraries and photon libraries.

Material compositions for the materials analyzed in this activation analysis were taken from the available literature. Material certifications for the Pathfinder reactor vessel and internals were unavailable and/or did not contain trace element data.

Based upon plant data, the Pathfinder plant operated for 16,635 MWd (thermal). Some discrepancies were encountered in the review of plant data regarding the plant's thermal output at 100% rated power. Early plant documents indicate a core design thermal output of 190 MWt, and a core design flux of 5.0 E+13 neutrons/cm²-sec. However, subsequent correspondence stated a thermal output of 199.6 MWt, and a design flux of 3.5 E+13 neutrons/cm²-sec over the life of the plant.

For input files used to analyze the activation of materials in the reactor core and at the reactor core boundary, a core average total flux of 5.0 ± 13 was used. After verification of the actual average core flux from the Pathfinder staff, the results generated by ORIGEN2 were scaled linearly down to correspond to the 3.5 ± 13 level, for the core and core boundary regions. This scaling was performed in the Lotus 1-2-3 spreadsheet, as given in Appendix B. All fluxes beyond the core boundary were used as given in the Pathfinder plant documents.

The total exposure (16,635 MWd) was used to provide the exposure time input necessary for ORIGEN2. To calculate the total effective full power days, the core design thermal output value of 190 MWt was used. This provided a conservative result of 87.6 effective full power days. Additionally, it was conservatively assumed that all of this exposure occurred immediately prior to plant shutdown.

The ORIGEN2 libraries were used as the source of neutron cross sections, photon information and decay data. For all calculations of neutron activation inside the reactor core, the ORIGEN2 BWR-U neutron cross section library was employed. For calculations of neutron activation outside the reactor core, the ORIGEN2 thermal library was used. The decay library used was the ORIGEN2-provided library. The H₂O bremsstrahlung photon library was used;

however, the choice of photon libraries has no impact on the results of concern in this analysis.

Figure A-1 contains a sample input file, and Figure A-2 a portion of the output file for the activation of stainless steel at the core boundary. This problem required a material composition for stainless steel and the thermal neutron cross section library as input.

Page 33 of 61

FIGURE A-1

ORIGEN2 SAMPLE INPUT

-1 .1 - 1 CALCULATION OF ACTIVATION AND CEPLETION IN THE PATHFINDER ATOMIC POWER PLANT INTERNALS, VESSEL, AND BIOLOGICAL SHIELD. THE PATHFINDER ATOMIC POWER PLANT HAD ITS INITIAL CRITICALITY ON MARCH 24, 1964. THE PLANT'S FINAL SHUTDOWN OCCURRED, DUE TO A CONDENSER TUBE FAILURE, ON SEPTEMBER 16, 1967. THE TOTAL CORE BURNUP AT THAT TIME WAS 16635 MMD. THE CORE DESIGN POWER IS 190 MMt. SINCE MOST OF THE POWER OPERATION OCCURRED AT THE END OF THE PLANT'S LIFE, WE WILL CONSERVATIVELY ASSUME THE PLANT OPERATED AT 100% POWER FOR 87.553 DAYS, UNTIL SEPTEMBER 16, 1964. USE DEFAULT FRACTIONAL RECOVERY VALUES, AS GIVEN BY FIRST THREE RECORDS OF THIS FILE. PROVIDE A TITLE FOR THE PROBLEM. TIT PATHFINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL. ٠ STORAGE VECTOR DEFINITIONS. -1 . STAINLESS STEEL. READ THERMAL CROSS SECTION LIBRARY AND PUT DATA ON UNIT 9. LIB 0 1 2 3 201 202 203 9 0 0 1 0 . READ H20-BREMSSTRAHLUNG PHOTON LIBRARY. PUT DATA ON UNIT 10. PHO 101 102 103 10 READ STAINLESS STEEL COMPOSITION. INP -1 1 4*0 . . ACTIVATION AND DEPLETION CALCULATION.

Page 34 of 61

FIGURE A-1

ORIGEN2 SAMPLE INPUT (CONTINUED)

```
PREPARE FOR ACTIVATION.
  BAS THERMAL KSEC - FLUX AT CORE BOUNDARY - STAINLESS STEEL.
    MOVE MATERIAL COMPOSITION TO OUTPUT VECTOR 1.
      -1 1 0 1.0
  NOV
    FLUX EXPOSURE OF MATERIAL OF INTEREST.
  BUP
        87.553 5.00E13 1 2 4 2
  IRF
 BUP
            DECAY OF THE IRRADIATED MATERIAL.
            PREPARE FOR THE DECAY CALCULATION.
.
.
    DECAY THE ACTIVATED MATERIAL OF INTEREST OUT TO JUNE 1, 1989.
.
 DEC 7929.0 2 3 4 1
.
    PROVIDE HEADINGS FOR THE COLUMNS.
     1 * STARTUP
 HED
      2 * SHUTI MN
3 *1-JUN-1989
 HED
 HED
.
   PRINT THE NUCLIDE AND CURIE TABLES ONLY, IN GRAMS, FOR THE MATERIAL
.
   OF INTEREST.
 OPTL
       6*8 5 17*8
 OPTA 24*8
 OPTE 24*8
 OUT
     3 1 6 0
 END
```

Page 35 of 61

FIGURE A-1

ORIGEN2 SAMPLE INPUT (CONTINUED)

:	MATERIA	L COMPOSI	TION.						
4	030000	1.30E-7	070000	4.52E-4	110000	9.70E-6	130000	1.008-4	
4	170000	7.00E-5	190000	3.00E-6	200000	1.90E-5	210000	3.00E-8	
4	220000	6.00E-4	230000	4.56E-4	240000	1.84E-1	250000	1.53E-2	
4	260000	7.06E-1	270000	1.412-3	280000	1.00E-1	290000	3.08E-3	
4	300000	4.57E-4	310000	1.296-4	330000	1.94E-4	340000	3.50E-5	
4	350000	2.00E-6	370000	1.00E-5	380000	2.00E-7	390000	5.00E-6	
4	400000	1.00E-5	410000	8.90E-5	420000	2.60E-2	470000	2.00E-6	
4	510000	1.23E - 5	050000	3.00E-7	560000	5.00E-4	570000	2.00E-7	
4	580000	3.71E-4	520000	1.00E-7	630000	2.00E-8	650000	4.70E-7	
4	660000	1.00E-6	670000	1.00E-6	700000	2.00E-6	710000	8.00E-7	
4	720000	2.00E-6	740000	1.86E-4	820000	6.70E-5	900000	1.00E-6	
4	920000	2.00E-6	0	0.0					
0				Martin States					

Page 36 of 61

FIGURE A-2

PARTIAL ORIGEN2 SAMPLE OUTPUT

		DUTIONT + 1	PAGE	13
	• P#	THE INDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL		
	PONE	R+ 0.000001+00 mm. BURNLP+ 0.0000001+00 mm0. FLLR+ 5.001+13 N/C+112-517	ACTIVATION PRODUCTS	
		7 MACLIDE TABLE RADIOACTIVITY. CURIES		
		THERMAL KELC . FLUX AT CORE BOLNDARY . STAINLESS STEEL		
		STARTUP SELITION 1- JUN- 1989		
*		0.0001+00 0.000+0000+00		
*	2	0.0001 +00 0.000+ +00 0.00+ +00		
H	3	0.000f+00 1.215f-05 3.593t-06		
*		0.0001+00 2.0501+12 0.0001+00		
et	3	0.000100.0000+000.000+00		
		0.0000+60 0.0000+00 0.0000+00		
ĸ		0.0001+00 0.0001+00 0.0001+00		
1		0.0001+00 0.0001+00 0.0001+00		
1		0.0301+00 0.0001+00 0.0001+00		
5		0.0001+00 5.2311+07 0.0001+00		
		0.0001+00 5.2310+07 0.0000+00		
	10			
	10			
	12			
č	12			
e	13			
c	14	0 0000 +00 1 374+04 1 370+04		
c	15	0.000f+00 1.792F+11 + 000F+00		
N	13	0.0001+00 0.0000+00 0.0001+00		
N	14			
N	15	0 0001+00 0 00+100 0 00+100		
N	16	0 0000 +00 2 3268 +09 0 0000 +00		
0	16	0.0001 000 000+ 1000 0 00+ 1000 00+ 1000 0		
0	17	0.000t+00 U.000t+00 0.000t+00		
0	18	0.0001 00 0 0001 00 0.000 00		
0	19	00001+00 0.0001+00 0.000+000		
1	19	0.0001+00 0.0002+00 0.000+00		
1	20	0.0001100.00000000000000000000000000000		
£ ()	20	0.0001+00.0.0002+00.0.000++00		
1	21	0.001 +00 0.001 +00 0.000 + 1000.0		
1	22	C 0001 +00 0 0001 +00 0 0001 +00		
	23	0.0001 +000 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
A :	22	0.0001 +00 0.0001 +00 0.0001 +00		
A :	23	0.0001 +00 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
A 1	24	0 0001+00 1 8191+04 0.0001+00		
A :	244	0.0062+00 1.3732+04 0.0002+00		
A .	25	0 0001+00 0.0001+00 0.0001+00		
	24	0.0001+00.0.0001+00.0.0001+00		
	25	0 0001+00 0 0001+00 0 0001+00		
	20	0.0001+00.0.0001+00.0.0001+00		
	24	0 0001+00 5 8071+19 0 0001+00		
	37	0 0001-00 0 0001-00 0 0001-00		
	28			
	29			
	30	0.0001+00 0.0001+00		
1	28	0.0001+00.0.0001+00.0.0001+00		
1	24	0.0001 +00.0 0.000 +000.0 0.000 +000		
1	00	6.000E+00 6.000E+00 0.000E+00		
11	-			

Page 37 of 61

FIGURE A-2

			A CONTRACTOR OF STATES AND STATES		PAGE 14
		* PATH	FINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL		
		-	0.000001+00 m AIRM P+ 0.000001+00 mm LUN- 1.001+13 MICHAEL	ACTIVA	TION PRODUCTS
0			7 MACLIDE TABLE : BADIDACTIVITY CURIES	arc	
199			THERMAL XSEC . FLUX AT CORE BOLNDARY . STAINLESS	STEEL	
			STARTUP SHLITDOWN 1- JUN- 1989		
51	9	11	0.0000+00 3.1866+17 0.0000+00		
51	3	12	0.000+1000.0 00+1000.0 00+1000.0		
	1	11	0.0001+00 0.0001+00 0.0001+00		
P	-	2	0.000E+00 4.270E-06 0.000E+00		
P	2	9	0.0001+00 2.1861-18 0.0001+00		
	1		0.0001+00 0.00(1+00 0.0001+00		
2	-	1			
š			0.0001+00 2.9521+04 0.0001+00		
š		6	0.0001+00 0.0001+00 0.0001+00		
5	3	7	0.0000+00 3.0766+11.0.0000+00		
CL	3	5	0.0001+00 0.0001+00 0.0001+00		
CL	3	6	0.0001+00 2.4591+08 2 4591+08		
CL	3	7	0 0001 +00 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
CL	3		0.0001+00 1.6651-04 0.0001+00		
CL	3	-	0.0001+00 1 9461-06 0 0001+00		
AR	3		0.0001+000 - 0001+00 0.0001+00		
AR	3	1	0 0001+00 7.6771-07 0.0001+00		
AR	3	•	0.0001 +00 0.00+ 1000.0 00+ 1000.0		
**	2	9	D.000f+00 1.444t-11 1.365t-11		
	2		0 0002+00 0 0000+00 0 0000+00		
	2		0.0001+00 1.4201-20 6.2424-24		
	5		0.0001+00 0.0001+00 0.0001+00		
	-	0	2 5716-15 1 7241-14 1 7241-14		
		1	0001+00 0.0001+00 0.0001+00		
		2	0.0001+00 6.1291-06 9.2671-21		
	4	3	0 0001+00 0 0001+00 0 0001+00		
		•	0 0001+00 0 00+1000 0 00+1000 0		
CA		0	0.0001+00 0.0001+00 0.0001+00		
CA	4	1	0.000f+00 3.067f-10 3.066f-10		
CA		2	0.0001+00 0.0001+00 0.0001+00		
CA	*	3	0.0001+00 0.0001+000 0.000+1000		
CA	*	11435	0.0001+00 0.0001+00 0.0001+00		
CA CA		2	0.0001+00 2.5051+06 5.6521+21		
~	-				
CA	-	1.11	0.0001+00 0.0001+00 0.0001+00		
CA		1.5.6.	0.000(+00 8.056(+07 0.000(+00		
SC			0 0001+00 0 0001+00 0 0001+00		
SC			0.0001+00 7 3631-06 0.0001+00		
SC			0 000f+00 5 164f-06 0 000f+00		
SC	47	1	0.0001+00 4 0211+08 0.0001+00		
SC	41		0001 +00 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
SC	4		0.0006+00 4.0568-07 0.0008+00		
SC	51		0001 100 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
	41		0001+00 0 0001+00 0 001+00		
71	51				
	1	1158			
	-	1.1.0			

Page 38 of 61

FIGURE A-2

		OUTPUT UNIT PAGE 15
		PATHE INDER ACTIVATION ANALYSIS. CLRIES PER GRAM OF MATERIAL
0		DETE + 0.000001+00 MM. BURNUP+ 0.000001+00 MMD. FLUX+ 5.001+3 N/CM++2-SEC 7 NUCLIDE TABLE: RADIDACTIVITY. CURIES
		THERMAL XSEC - FLUX AT CORE BOLADARY - STAINLESS STEEL.
		STARTOP SECTORES 1-JUN-1889
	54	1 9997-10 1 9447-10 0 0437-10
	51	0.001 +00 0.00+ 100 0.00 +00
•		6.0001+00 3.7551+02 0.0001+00
۲	63	0.0001.00.0.0001+00.0.0001+00
۲	54	00+1000.0 00+1000.0 00+1000.0
CA	56	0 00C/+00 0 0000+00 0 0000+00
CR	51	0 0001-00 1 7601-00 6 0001-00
c		6 000 +00 2 5(7) 02 0 000 +00
-	54	0 0001 +00 0 0001 +00 0 0001 +00
-		0 0001+00 0 0001+00 0 0001+00
*	50	6.0001+00 2.9971+00 0.0001+60
*	\$7	0.0011000.000+1000.00+1000
*	50	0 0001 00 0 0001 00 0001 00
	54	0 0001+00 0 0001+00 0 0001+00
	30	0.000/100 8.3201-02 2.5501-04
	57	
	50	
	54	0 0001+00 2 5491+02 0 0001+00
co	50	0.000100.000100.000+000
co	50	0.0011000.0 00+1000.0 00+1000
CO	59	00+1000.0 00+1000.0 00+1000.0
co		© 006f+00 2.215f+02 1.274f+03
co	00	0.0001+00 3.4341+01 0.0001+00
0		0.0007400 5.3241404 0.0007400
NI	50	
NI	59	
NI	60	0,9001-00,0,001-00,0,0001-00
-		0.0001 00 0.0001 00 0.0001 000
NI	42	00+1000.0 00+1000.0 00+1000.0
NI	63	0.000f+00 1.266f+03 1.075f+03
NI	84	0.0001+00 0.0001+00 0.0001+00
NI	65	0.0006+00 1.6828-02 0.0008+00
NI		0.0001+00 2.4938+07 0.0001+00
0		
eu		
cu	65	0 0001+00 0 0008+60 0 0008+60
cu		0.0001+30 2.6341+02 0.0001+00
cu	67	D. 0001+00 7 6491-08 D. 0002+00
ZN	6.3	000100000000000000000000000000000000000
ZN	64	0.0061+00.0.0001+00.0001+00
IN	*5	0.0601+00 4 7611-04 7.7631-14
ZN	86	0.0001+00 0.0001+00 0.0001+00
ZN		0.0001400.0.0001400
	-	
		사람 방법은 경험을 많은 것을 물건이 가격 전화에 있는 것을 것이 하는 것을 것 같아요. 같은 것을 것 같아요. 그는 것 같아요. 그는 것 같아요. 그는 것 같아요. 것 같아요. 것을 같이 것 같아요. 것을 같이 것

Page 39 of 61

FIGURE A-2

••••••••••••••••••••••••••••••••••••	1				PAGE 16
Reflet 0 00000101 MERLEY 0 DESCRIPTION REFLET 0 DESCRIPTION		. PAT	WINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL.		
P PARAL DI TABLI - BADICACTIVIT, CARIS' UNDERAL XEC TAUX AL CORT ACADAY - STAIRE SES STEL STATUS - BUTTORY 1-AN-1980 DA SOULO DI LATTA S LOBOLEO DA TO DO SOULO DI LATTA S LOBOLEO CA TO DO SOULO DI LATTA S LOBOLEO		POWER	0.000000 -00 M BURNER 0.000000 -00 MM 5118- 5.001-11 N/CH112	ACTIVATIO	N PRODUCTS
THREAL SOLG - FLUX AT CORE BOLADARY - STAINEISS STEL. D1 64 6 0001100 1 140113 6 0001100 D1 64 6 0001100 1 140113 6 0001100 D1 64 6 0001100 1 140113 6 0001100 D1 74 6 0001100 1 100110 1 1001100 D1 74 6 0001100 1 1001100 D1 74 6 0001100 1 100110 1 1000100 D1 74 6 0001100 1 1001100 D1 74 6 0001100 1 1001100 D1 74 6 0001100 1 10011000 D1 74 6 0001100 1 10011000 D1 74 6 0001100 1 100011000 D1 74 0 0001100 1 1000000000 </th <th>0</th> <th></th> <th>7 MULIDE TABLE : RADIDACTIVITY. CURIES</th> <th>arc</th> <th></th>	0		7 MULIDE TABLE : RADIDACTIVITY. CURIES	arc	
11 40 0 100110 1 - (x + 1) 400110 12 40 0 100110 1 - (x + 1) 400110 0 100110 13 7 0 100110 1 - (x + 1) 400110 0 100110 14 7 0 100110 1 - (x + 1) 400110 1 100110 1 100110 14 7 0 100110 1 - (x + 1) 400110 1 100110 1 100110 1 100110 14 7 0 100110 1 - (x + 1) 400110 1 100110 1 100110 1 100110 1 100110 1 100110 14 7 0 100110 1 - (x + 1) 400110 1 1001100 1 10011000 1 1000			THERMAL KSEC . FLUX AT CORE BOLADARY - STAINLES	STEEL	
A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A Her C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A T C BORT-RD / 144 - 2 G BORT-RD A			STARTUP SPLITDOWN 1- JUN- 1639		
1/1 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0	20		0.0001+00 1.1441+03 0.0001+00		
A 7 0 0 0 0 0 CA 8 0 0 0 0 0 0 0 CA 8 0 0 0 0 0 0 0 0 CA 7 0	ZN	70	0.0001+00 0.0001+00 0.0001+00		
2H 74 0.0001+00 0.0001+00 0.0001+00 CA 70 0.0001+00 0.0001+00 0.0001+00 CA 72 0.0001+00 0.0001+00 0.0001+00 CA 73 0.0001+00 0.0001+00 0.0001+00 CA 74 0.0001+00 0.0001+00 0.0001+00 CA 77 0.0001+00 0.0001+00	ZN	71	0.0001+00 2.4251-06 0.0001+00		
CA 0 -OCI-00 0.000-00 0.000-00 CA 7 0 -OCI-00 0.000-00 0.000-00 CI 7 0 -OCI-00 0.000-00 0.000-00 CI 7 0 -OCI-00 0.000-00 0.000-00 CI 7 0 -OCI-00 0.000+00 0.000+00 CI 7 0	ZN	7 14	0.0001+00 3.0661+07 0.0001+00		
G. 76 0.001+00 1.001+00 0.001+00 G. 72 0.001+00 1.001+00 0.001+00 G. 73 0.001+00 1.001+00 0.001+00 G. 74 0.001+00 1.001+00 0.001+00 G. 74 0.001+00 1.001+00 1.001+00 G. 74 0.001+00 1.001+00 0.001+00 G. 74 0.001+00 0.001+00 0.001+00 G. 75 0.001+00 0.001+00 0.001+00 G. 75 0.001+00 0.001+00 0.001+00 G. 75 0.001+00 0.001+00 0.001+00 G. 76 0.001+00 0.001+00 0.001+00 G. 77 0.001+00 0.001+00 0.001+00 <td< th=""><th>GA</th><th>69</th><th>00+1000.0 00+1000.0 00+1000.0 00+1000.0</th><th></th><th></th></td<>	GA	69	00+1000.0 00+1000.0 00+1000.0 00+1000.0		
GA 71 0 0001-00 0 0001-00 0 0001-00 0 GA 72 6 0001-00 0 001-00 0 0001-00 GA 72 6 0001-00 0 0001-00 0 GT 70 0 0001-00 0 0001-00 0 GT 71 0 0 0001-00 0 0001-00 0 GT 72 0 0 0001-00 0 0001-00 0 GT 72 0 0 0001-00 0 0001-00 0 GT 72 0 0 0001-00 0 0001-00 0 GT 74 0 0 0001-00 0 0001-00 0 GT 74 0 0 001-00 0 0001-00 0 GT 74 0 0 001-00 0 0001-00 0 GT 74 0 0 001-00 0 0001-00 0 GT 75 0 0 001-00 0 0001-00 0 GT 74 0 0 001-00 0 0001-00 0 GT 75 0 0 001-00 0 0 0001-00 0 GT 75 0 0 0001-00 0 0 0001-00 0 GT 75 0 0 001-00 0 0 0001-00 0 GT 75 0 0 001	GA	70	0.0001+00 1 5191-03 0.000E+00		
GA 72 G. 0001-00 2.513-63 G. 0001-00 GI 70 G. 0001-00 1.5071-07 G. 0001-00 GI 71 G. 0001-00 1.5071-07 G. 0001-00 GI 73 G. 0001-00 1.5071-07 G. 0001-00 GI 74 G. 0001-00 1.5071-07 G. 0001-00 GI 75 G. 0001-00 1.5071-07 G. 0001-00 GI 76 G. 0001-00 1.5071-07 G. 0001-00 GI 77 G. 0001-00 1.5071-07 G. 0001-00 GI 78 G. 0001-00 1.5071-07 G. 0001-00 GI 78 G. 0001-00 1.5071-07 G. 0001-00 GI 79 G. 0001-00 1.5071-07 G. 0001-00 GI 70 G. 0001-00 1.5071-07 G. 000	GA	71	0.0008+00 0.0008+00 0.0008+00		
CA 72 C 0001-00 0 0001-00 0 0001-00 Cf 71 0 0001-00 0 0001-00 Cf 72 0 0001-00 0 0001-00 Cf 72 0 0001-00 0 0001-00 Cf 72 0 0001-00 0 0001-00 Cf 74 0 0001-00 0 0001-00 Cf 74 0 0001-00 0 0001-00 Cf 74 0 0001-00 0 0001-00 Cf 77 0 000	GA	72	0.0006+00 2.9136-03 0.0006+00		
11 0 0001+00 1 0001+00 1 0001+00 1 0001+00 0 0 <	GA	724	D.000t+00 9.001t+05 0.000t+00		
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KR 80 0.000f+00 0.000f+00 c.000f+00 KR 81 0.000f+00 2.194f+13 2.193f+13 KR 81M 0.000f+00 1.798f+07 0.000f+00 KR 82 0.000f+00 0.000f+00 0.000f+00	-	7 94	0.9001 +00 0 0001 900 0 0001 900		
KK 81 0.0000000 2.1940-13 2.1930-13 KK 81K 0.0000000 1.7980-07 KK 82 0.0000000 0.0000000	KR	.0	0.0001+00 0.0001+00 0.0001+00		
KK 81A 0.000E+00 1.756E+07 0.000E+00 KR 82 0.000E+00 0.000E+00 0.000E+00	**		0.0008+00 2.1948-13 2.1938-13		
NK 82 0.0000+00 0.0000+00 0.0000+00	KR		0 0008+00 1 7441-07 0 0001+00		
	AR	82	0.0001+00 0.0001+00 0.0001+00		

Page 40 of 61

FIGURE A-2

•		OUTPUT UNIT		PAGE 17
	• PAT	THE INDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL.		
	POWER	* 0.000000+00 MM. BURNUP+ 0.000000+00 MMD. FLUX+ 5.000+13 N/CA+12-SEC	ACTIVATION PRO	ALC IS
0		7 MACLIDE TABLE: RADIOACTIVITY, CURIES		
		THERMAL XSEC . FLUX AT CORE BOLNDARY . STAINLESS STEEL		
	25.33	STARTUP SHUTDOWN 1- JUN- 1989		
KR		0.0001+00 0.0001+00 0.0001+00		
		0.000t+00 1.643t+06 0.000t+00		
KR		0.000f+00 2.047f+12.0.000f+00		
-		0.0001+00 0.0001+00 0.0001+00		
	67	0.0001+00 1.5971-21 0 0001+00		
		0.0001+00 3.1641-25 0.0001+00		
	45	0.0001+00 0.0001+00 0.000+000		
RB		0.0001+00 3.0371+05 0.0001+00		
		0.0001+00 3.4331+06 0.0001+00		
-				
RB	89	0.0001+00 2 4441+13 0 0001+00		
SR		0.0001+00 0.0001+00 0.0001+00		
SR	45	0.0001+00 4.6281-09 0.0001+00		
SR		0.0001+00 5.7161-09 0.0001+00		
58	46	0.0001+00 0.0001+00 0.0001+00		
SR	87	0.0001+00.0.0001+00.0.0001+00		
SR	874	0.000f+00 1.005f+07 0.000f+00		
SR		0.0001+00 0.0001+00 0.0001+00		
58	90	0.0001+00 0.7241+09 0.0001+00		
58	91	0.000(+00 1.670(-16.0.000(+00		
SR	93	0.0001+00 0.0001+00 0.0001+00		
Y		0.0001+00 0.0001+00 0.0000+00		
¥		0.0001 +00 0.0001 +00 0.0000 +000		
Y	90	0.0008+00 5.4518-05 1.6708-15		
4	904	0.0001+00 4.5721-04 0.0002+00		
Y	91	00+100.0 0.1100.0 0.0001.0		
1	92	0.0000+00 1.9921+'3 0.0000+00		
2	*3	0.000t+00 0.000t+00 0.000t+00		
4				
ZR		0.0001+00 0.0001+00 0.0001+00		
28	90	0.0001+00 0.0001+00 0.0001+00		
ZR	91	0 0001 00 0 0001 00 0 0001 00		
ZR	92	0. 1000 + 100 0. 1000 + 1000 0 00 + 1000 0 00 + 1000 0		
28	93	0.0008+00 4.3038-13 4.3038-13		
ZR	94	0.000+300.0 0.00+3000.0 0.00+3000.0		
28	*5	0.0000+00 5.3221-07 0 0000+00		
28		0.0001+00 4.3441-04 0.0001+00		
-		0.000 +00 0.000 +00 0.000 +00		
-	2	0 0000 +00 0 0000 +00 0 0000 +00		
-	2	0.0001+00 0.0001+00 0.0001+00		
-	M	9.0001+00 2.4471+15 2.7441+13		
-	•	0.0008+00 7.3128+09 7.3068+09		
-	15	0.0000+00 2.7171-06 0.0000+00		
-	54	0.0001+00 3.7261+09 0.0001+00		
-	17	0 0001+00 4 1671+04 0 0001+00		
1	1	0.0001+00 4.7441-08 0.0001+00		

Page 41 of 61

FIGURE A-2

•		TPUT UNIT	PAGE 18
	PATHE INDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL		
PO	WER. 0.000000+00 MW. BURNUP+ 0.000001+00 MMD. FLUX+ 5.000+13 N/CM+12-SEC	ACTIVATION PEO	DUCTS
0	7 MUCLIDE TABLE: RADIDACTIVITY. CURIES		
	THERMAL XSEC . FLUX AT CORE BOUNDARY - STAINLESS ST	TEL	
	STARTUP SHUTDOWN 1- JUN-1989		
NB 978	0 0000+00 4 0161-08 0 0000+00		
NO 98	0.0001+00 0.0001+00 0.0001+00		
	0 0001+00 1.9571+04 0.0001+00		
	0.0001+00 7.6981-06 7.8681-08		
.0 94	00+1000.0 00+1000.0 00+1000.0		
80 95	00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0		
MO 96	00+1000.0 00+1000.0 00+1000.0		
MO 97	00+ 1000 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
A0 96	0 0001 100 0 0001 100 0 0001 0 000 000		
-0100			
-0101	0.0001+00 4.2111+03 0.0001+00		
10 97	0.0001+00 0.0001+00 0 0001+00		
10 974	00+ 1000 0 00+ 1000 0 00+ 1000 0		
TC 98	00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0		
TC 99	0.0000+00 6.3812-10 6.6852-10		
TC 100	0.0001+00 5.8791-06 0.0001+00		
TCIDI	0.0001+00 4.2111-03 0.0001+00		
RU 90	0.0001-00.0.0001-00.0.0001-00		
	0.0001+00 0.000+000 0.0000+000		
	0.0001+000 0 0001+000 0 00+1000 0 00+1000 0		
-	0.0001+000 0.0001 0 00+1000 0 00+1000.0		
RU102	0.0001+00 0.000+000 0.0001+00		
RUIDS	0.0008+00 4.4238-10 0.0008+00		
RUIDA	0.0000000000000000000000000000000000000		
RUIGS	0.0001+60 1.1021+19 0.0000+00		
81/107			
RH102	0.0000 +00 0.0000 +01 0.0000 +00		
KH103	0.0300+00 0.0000+10 0.0000+00		
RH104	0.0001+00 6 5782 13 0.0001+00		
-	0.0008+00 4.0208-14 0.0008+00		
RH105	0.0000+00 7.9771-19 0.0000+00		
RH1054	0.0001+00 3.0661-20 0.0001+00		
RH106	0.0001+00 1.1171+20 0.0001+00		
RH107	6 0001+00 0 0001+00 0 0001+00		
PD102	0.000f+00 0.000f+00 0.000f+00		
PDIDO	0 0001 +00 0 0001 +00 0 0001 +000		
PDIDA	00+ 3000. 0 00+ 3000. 0 00+ 3000. 0		
PD105	0.0001 +00 0 0001 +00 0 0001 +000		
PD 106	00+1000 0 00+1000 0 00+1000 0 00+1000		
PD107	0.0001/00 0.0001+00 0.00001+00		
PDIDA			
PDIOS	0 0001+00 2 4741+04 0 0000+00		
PD1094	0.0001+00 4.7215-10 0.0001+00		

Page 42 of 61

FIGURE A-2

	OUTFUT UNIT . 8 PAGE 19
	PATHE INDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL.
	ACTIVATION PRODUCTS
0	2 NET 0 00000100 mm 0 00000100 mm0. FLUX 5 00F+13 N Ca++2-SEC
	THE ALL STALL SADDACTIVITY, CARTES
	STARTUP SHATTORN 1- IN- 1989
PDIII	0.0000+000 1.5400+10 0.0000+00
POIL	6 0.000E+00 1.442E+11 0.000E+00
AC106	0.0001 00 0001 00 0000 00
AG107	6.0001 0001 00 0000 00 0000 00
AC108	6.000t+00 2.636t+04 2.405t+09
ACIDA	0.000E+00 3.042E+08 2.702E+05
AG109	¢.000E+20 0.000E+00 0.000E+00
AG109	6 0000+00 3.502t-08 4.491E-14
AGI10	0.0001+00 6.2421-04 2.5071-17
ACTIO	0.0000 00 6.7181-06 1.8851-15
AGIII	0.0001-00 1.7161-06 0.0001-00
10112	
CD105	0.0001-00.0.001-00.0.001-000
CD107	
CDIDA	
CDIDS	
CDIIO	0 5001-00 0 0001-00 6 0001-00
CDIII	0.0001 +00 0.0001 +00 0.0001 +00
CDITI	0.0001+00 2.4061+04 0.0001+00
CD112	0.0000 + 000 0 + 0000 + 00
CDIIS	0.0006+00.0.0006+00.0.0006+00
CD114	0 0000 0 000 + 000 0 0000 + 000
CD115	0.000f+00 1.395f-16 0.000f+00
CD1150	0.000[+00 3.422[-14 0.000[+00
CDII6	0.000f+00 0.000f+00 0.000f+00
CD117	0.0001 + 000 0.0001 + 000 + 000
CD117#	P. 0001 + 00 0.0001 + 00 0.0001 + 00
CDITY	0.0001+00 0.0001+00 0.0001+00
INTIA	
10113	
INIIA	
INIIS	0 0001+00 2 5421-14 1 0001-33
IN116	0.0001+00 1.3011-20 0.0001+00
	0.0001+00 4.5391-20 0.000+00
IN117	0.004 1000.0 00+ 1000.0 00+ 1000.0
IN117#	0.00+1000.0.00+1000.0.00+1000.0
18316	000+1000.00+1000.0
IN119	0.0001 + 00 0.0001 + 00 0.0001 + 00
111194	0.004 100 0.004 100 0.004 100
IN120	0.0001+00 0.0001+00 0.0001+00
IN1204	0.0001+00 0.0001+00 0.0001+00
64113	0.0001+00.0.0001+00.0.0001+00
SNILL	
501134	
Shill	
SHIIS	
5N116	0.0001+00 0.0001+00
SN117	0.0001+00.0.001+00.0.0001+00
	1997년 1998년 2월 1998년 1998년 1997년 1997년 1977년 1978년 1

Page 43 of 61

FIGURE A-2

1		OUTPUT UNIT	PAGE 20
• •	ATHFINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL		
PON	18: 0 000001+00 mm. BLENLE: 0 000001+00 mm. 110: 5 001+13 H/(****	ACTIVATION	PRODUCTS
0	7 MALIDE TABLE MADIOACT VITY, CURIES	1.910	
	THERMAL XSEC . FLUX AT CORE BOLNDARY - STAINLE	55 STEEL	
51.12.63	STARTUP SHUTDOWN 1- JUN- 1989		
5N117#	0.0008+00 0 0008+00 0 00+3000 0 00+3000		
51118	0.000+3000.0 00+3000.0 00+3000.0		
511119	0.0001+00 0.0001+00 0.0001+00		
54170	0.0001+00 0.0001+00 0.0001+00		
SN121			
5N1218	0.0001+00 0.0001+00 0.0001+00		
SN122	0.0001 +00.0.0001 +00.0.0001 +000		
SN123	0.0002+00 4.2652-11 3.0621-24		
	0.0000+00 2.5400-12 0.0000+00		
SN124	00+ 1000 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
SN125	0.000 + 1000. 0 00+ 1000. 0 00+ 1000. 0		
SN1254	0 0001 +00 0 0001 +00 0 0001 +00		
58121	0.0001+000 0 00+000 0 00+000 0 00+1000 0		
58122	0.0001+00 2.9391+04 0.0001+00		
501224	0 0001+00 2.5661+06 0.0001+00		
58124	0.0001+00 0.0001+00 0.0001+00		
581248	0.0001+00 0.1471+07 0.0001+00		
58125	0.0001+00 8.0261-09 3.5116-11		
58126	0.0000+00 3.1630+11 0.0000+00		
581268	0.0000+00 3.9471-11 0.0000+00		
16130	0.0001+00 0.0001+00 0.0001+00		
11121	0.000f+00 0.000f+00 0.000f+000		
TE 121#	00+1000.0 00+1000.0 00+1000.0 00+1000.0		
16122	0.000+1000.0 00+1000.0 00+1000.0		
16123	0.0000 +00 1.5301-21 2.3361-21		
101230	0 0001+00 2 4611+08 2 7971+28		
TEIDS			
TE1254	0 0001+00 7 4941-10 4 5461-12		
71 126	0.0001 +00 0.000 + 100 0.000 + 1000 0.00 + 1000 0.0		
16 127	0.0001+00 1.0521-14 0.0001+00		
TE 127#	0.0001+00 1.8461-16 0.0001+00		
TE 120	0.0001+00 0.0001+00 0.0001+00		
11129	0.0001+00 0.0001+00 0.0001+00		
TE 1 1944	0.0001+00 0.0001+00 0.0001+00		
11130	0.0001+00 0.0001+00 0.0001+00		
11131			
1125			
1126	0 000[+00 0 000[+00 0 000]+00		
1127	0.0001+00 0.0001+00 0.000100		
1126	0.0002+00 5.7842-18 0.0002+00		
1129	00+ 1000 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
1130	0.0002+00 0.0002+00 0.0002+00		
1130	0+1000.0 00+1000.0 00+1000.0		
1131	0.0001+00 0.0001+00 0.0001+00		
1132	000110000110000100000000000000000000000		
1 1 24	0.0001+00 0.0001+00 0.0001+00		
11254	0.0001+00 0.0001+00 0.0001+00		
	v. 0001.00 0.0001.00		

Page 44 of 61

FIGURE A-2

1	-		PAGE	21
	ATHFINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL			
		ACTIVATION PRODUCTS	6 28	
0	THE D. DOUDER OF BORNER O. DOUDOFFOD MED. FLORE 5. DOE+13 N/CM++2-SEC			
1000	THERMAL XSIC - FLIX AT COST BOX MOUNT - ETAIN SEE ET			
	STARTUP SHUTDOWN 1- JUN- 1989			
XE 126	0. 0001 +00 0. 0001 +00 0. 0001 +00			
X£ 127	00+ 1000.0 00+ 1000.0 00+ 1000.0 00+ 1000.0			
XI 127#	00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0 00+ 1000 . 0			
XE 128	00+1000.0 00+1000.0 00+1000.0			
XE 129	0.0001+00 0.0001+00 0.0001+00			
NE 1294	0.0001+00 1.5821+22 0.0000+00			
XE 1.30	0.000f+00 0.000f+00 0.000f+00			
XE 131	0.000 +00 0.000 +00 0.000 +00			
X2 13 10	0.0001+00 0.0001+00 0.0'01+00			
At 132	0.0001+00 0.0002+00 0.0001+00			
AC133	0.0001+00 5.1291-11 0.0001+00			
AL IJA	0.0001+00 3.3501+12 0.0001+00			
XE 135	0.0001+00 9.2911-19.0.0001+00			
XE 1350	0.0001100 4.5251-20.0.0001400			
XE 136	0.0001+00 0.0001+00 0.0001+00			
XE 137	0.0000+00 4.143(-23.0.0000+00			
C5131	0 0000+00 4 2791-05 0 0000+00			
C5132	0.0001+00 0.0001+00 0.0001+00			
C5133	0.0001+00 0.0001+00 0.0001+00			
C5134	0.0001+00 3.9451-06 2.6721-09			
C5134#	0.0008+00 4.5418-06 0.0008+00			
C\$135	0 0006+00 9 6206-14 9 6206-14			
CS136	0.0001+00 2.9161+09 0.0001+60			
C5137	0.0006+00 5.0028-16 3.0298-16			819
C5138	0.0001+00 3.7581-18 0.0001+00			12 5 13 5 5
BA 130	0 0001+000 0 00+1000 0 00+1000 0			11 S. M. P. 19
BAIJI	0.000t+00 4.355t+05 0.000f+00			
BAIJIA	0 0001+00 & 1051+06 0.0001+00			1.0
BA132	0 0001+00 0 0001+00 0 0001+00			
BA133	0.0001+00 4.1541-07 1.0272+07			11-1-07-1
BALLA				
84135				
84135#	0.0001+00 1 1221-05 0 0001+00			
8A136	0 0001+00 0 0001+00 0 0001+00			
BA1.368	0.0000+00 2.7138-06 0.0000+00			al Kettin
SA 137	0.0000 +00 0.0600 +00 0.0000 +00			1.
84137M	0.0006+00 2.3448+06 2.6368+16			11 Jan 19
BA138	0.0008+00 0.0008+00 0.0008+00			1.5.1.1
BA139	0.000E+00 7.435E+04 0.000E+00			
BA 140	0.0000+00 1.5411-09 0.0000+00			
BAIAI	0.0001+00 2.0161+13 0.0001+00			
LA137	0.0000+00 8.1321+11 8.2062+11			
LA138	3.3436+18 3.1746+18 3.1746+18			Market .
14140	0.0001+00 0.0001+00 0.0001+00			
LAIAL				
CELIN				
CE 132	0 0001+00 2 9551-05 0 0001+00			
CE 137#	0.000(+00) 4771-06 0.000(+00			
CE 136	0.0001+00 0.0000+00 0.000+00			
and the second second				G

Page 45 of 61

FIGURE A-2

1					PAGE	22
	• PAT	WINDER ACTIVATION ANALYSIS. CURIES PER GRAN OF MATERIAL.				
	PONER	0.000005+00 MR. BURNER 0.000001+00 MR. TUN. 5.001+13 M/CH113-1		ACTIVATION PRODUCTS		
0		7 MACLIDE TABLE BADIDACTIVITY, CURIES				
		THERMAL XSEC + FLUX AT CORE BOLADARY + STAINLESS	STEEL			
		STARTUP SHUTDOWN 1- JUN- 1989				
CE	139	0.0008+00 2.1418-06 9.6878-24				
CI	1394	0.000E+00 8.102E+08 0.000E+00				
CI	140	0.0001+00 0.000+00 0.0000+00				
ct	141	0.0001+00 9.1512-04 0.0001+00				
CE	147	9.994E-13 9.990E-13 9.990E-13				
	143	0 0001+00 2 2671+04 0.0001+00				
		0.0001-00 2.1901-09 8.7421-18				
PP						
PR	147	0.0001400 2.5731.04 0.0001400				
PR	1424	0 0001 +00 \$ 7281 -07 0 0001 +00				
PRI	143	0.0000 +00 2.1756 +04 0.0000 +00				
PRI	144	0.0000 +00 1.6341-06 8.6771-18				
PRI	1+5	0.0001+00 3.8791+12 0.0001+00				
NDI	142	0001+000 0 000+1000.0 00+1000 9 00+1000 0				
NDI	14.3	00+1000.0 00+1000.0 00+1000.0 00+1000.0				
NOI	144	0.0001+00 7.8221-22 7.8311-22				
NO	145	0.0000 +00 0.0001 +00 0.0001 +00				
NO	146	0.01 36+00 0.0006+00 0.0001+00				
NO	147	0.000E+00 1.168E+13 0 000E+00				
NOT	40	0.0401+00 0.0000+00 0.0000+00				
NDI		0.0001+00.0.0001+00.0.0001+00				
-	150					
-	44					
Pal	47	0 0001+00 5 5795-15 2 2431-17				
PAI	48	0.0001+00 2.2161-14 0.0001+00				
-		0.0001+00 4.2511-16 0.0001+00				
PAI	49	0.0001+00 3.4171-15 0.0001+00				
PMI	50	0.0001+00 7.3666-17 0.0008+00				
PMI	51	0.0001+00 0.0001+00 0.0001+000 0				
Pat	52	0.0001+00.0.0001+00.0.0001+000.0				
54.1	44	0.0001+00 0.0001+00 0.0001+00				
54.	**	0.0008+00 1.4618-09 1.6008-16				
541	46	0.0001+00 5.3951+19 5.3951+19				
2411	•7	3.3551-16 3.2741-16 3.2741-16				
3411	40	J. 3561-21 3.461E-21 3.461E-21				
541	10	5.30/1-21 1.632t-25 1.632t-25				
541	51	0.0001400 0.0001400 0.0001400				
-	52	0.0001+00 0.0001+00 0.0001+00				
	53	0.0001+00 2 9221+05 0 0001+00				
541	54	0.0001+00 0.0001+00 0.0001+00				
-	55	0 0006+00 6.7138+07 0.0008+00				
LUI	51	0.0001+00 0.0001+000.0 00+1000.0				
EUI	52	0.000E+00 5.451E+07 1.803E-07				
LUI	524	0.0000+00 5.3351-06 0.0001+00				
tun	53	0.0001+000.0.0001+000.0.0001+000				
tuit	54	0.6001+00 3.6701-07 6.3801-06				
IUI	55	0.0001+00 1.2671+07 6.1921-09				
tun	96	0 0001+00 4 0261-06 0 0001+00				
001	24	0.0002+00 3.8692+20 5.1532+20				1.3.3.3.3

Page 46 of 61

FIGURE A-2

INDEC ACTIVATION AN 0.000002+00 am. BL 7 NLCLIDE 5TARTUP SHUTDOWN 0.000+00 2.023+00 0.000+00 2.023+00 0.000+00 3.504+12 0.000+00 3.504+12 0.000+00 0.000+00 0.000+00 0.000+00 0.000+00 0.000+00 0.000+00 0.000+00 0.000+00 4.768+05	ALVSIS. CURIES PER GRAB OF BATERIAL. RNLF+ D. DODODE+00 BMD. FLUX+ 5.00E+13 N/CH++2 TABLE: RADIOACTIVITY. CURIES HEBMAL XSEC - FLUX AT CORE BOUNDARY - STAINLES 1 - JUN-1089 2.776E-16 0.000E+00 0.000E+00 0.000E+00 0.000E+00 0.000E+00 0.000E+00	ACTIVATION PRODUCTS	
0.00002000 mm. BL 7 NLCL 100 5TARTUP SHUTCHIN 0.0000 +00 2.0231-000 0.0000 +00 2.0231-000 0.0000 +00 0.0000 +00 0.0000 +00 0.0000 +00	RNLP+ 0.00000E+00 MMD. FLUX+ 5.00E+13 N/C#**2 TABLE: RADIOACTIVITY, CURIES HERMAL XSEC - FLUX AT CORE BOUNDARY - STAINLES 1 - JUN-1989 2.776E+16 0.000E+00 0.000E+00 0.000E+00 0.000E+00 0.000E+00	ACTIVATION PRODUCTS	
7 NLCL IDE 7 NLCL IDE 7 NLCL IDE 5TARTUP SHL/TDOMA 0.0001+00 2.0231-06 0.0001+00 2.0201-00 0.0001+00 3.504E+12 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 4.768E-05	MALE 0 DODOF+00 AND PLOK 5.002+13 N/CA*2 TABLE: RADIOACTIVITY, CURIES HERMAL XSEC - FLUX AT CORE BOUNDARY - STAINLES 1 - JUN-1989 2.776E+16 0.000E+00 0.000E+00 0.000E+00 0.000E+00 0.000E+00 0.000E+00	IS STEEL.	
5TARTUP SHUTDOWN 0006+00 2 0231-06 0006+00 0 0000+00 0006+00 0 0000+00 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 4 7681-05	HRBAL SEC - FLUX AT CORE BOLNDARY - STAINLES 1 - JUN- 1989 2 .776E - 16 0 .000E+00 0 .000E+00 0 .000E+00 0 .000E+00 0 .000E+00 0 .000E+00	IS STEEL.	
STARTUP SHUTCHM 00001+00 2 0231-06 00001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 4 7681-05	1 - JUA- 1984 2 .7761 - 16 0 .0001 +00 0 .0001 +00 0 .0001 +00 0 .0001 +00 0 .0001 +00 0 .0001 +00	a sint.	
00001+00 2.0231+06 00001+00 0.0001+00 00001+00 3.5061+02 00001+00 0.0001+00 00001+00 0.0001+00 00001+00 0.0001+00 00001+00 0.0001+00	2.7761-16 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00		
0001+00 0 0001+00 0001+00 3 5061+12 00001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 4 7681-05	0.000t+00 0.000t+00 0.000t+00 0.000t+00 0.000t+00		
0000+00 3 506+12 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 0 0000+00 0000+00 4 768-05	0.000t+00 0.000t+00 0.000t+00 0.000t+00		
0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 4 7681-05	000+1000.0 000+1000.0 0.0001-00		
000f+00 0.000f+00 000f+00 7.000f+00 000f+00 0.000f+00 000f+00 4.768F-05	0.000f+00 0.000f+00		
0001+00 0.0001+00 0001+00 0.0001+00 0001+00 4.7681-05	0.0001+00		
000E+00 0.000E+00 000E+00 4.768E-05			
.000E+00 4.768E-05	0.0001+00		
the second se	0.0001+00		
00+ 1000 . 0 00+ 1000	0 0001+00		
.000E+00 7.156E-15	0.0061+00		
.000E+00 3.552E-18	0.0001+00		
0001+00 1 0316-10	9.3741-11		
0001+000 0 0001+00	0.0001+00		
0002+00 3.1671-05	0.000(+00		
0001+00 7.5341+60	0.0001+00		
0001-00 1 1441-10	0.0001+00		
0001+00 0.0001+00	0.0001+00		
0001+00 4.3041-00	0.0001+00		
0001+00 7 5635-04	2.0101-24		
0001+00 0 0001+00	0.0001+00		
0401+00 0 0001+00	0.000(+00		
0001+00 0 0001+00	0.0001+00		
.0001+000 0 0001+00	0.0001+00		
00+ 1000 . 0 00+ 1000	0.0001+00		
000E+00 1.547E-03	0.0001+00		
.0001+00 9.2781-04	0.0001+00		
0006+00 3.6818-06	0.0001+00		
00+ 1000 0 00+ 1000.	0.0001+00		
00+ 1000 . 0 00+ 1000	0.0001+00		
.000E+00 3.641E-04	0.0001+00		
000E+00 2.617E-09	2.5641-09		
00+ 1000 0 00+ 1000	0.0001+00		
00+ 1000 0 00+ 1000	0.0001+00		
0001+00 0 0001+00	0 0001 +00		
11-1006+00 1.3061-11	0.0008+00		
0001+00 0.0001+00	0.0001+00		
00+ 1000 0 00+ 1000	0.0000 +00		
0008+00 1.9151+06	0.0001+00		
0008+000 0 0008+000	0.0001+00		
000E+00 1.458E-10	0.0001+00		
000E+00 0.000E+00	0.0001+00		
0001+00 5.4758-11	0.000E+00		
0002+00 3.2122-14	0.000000		
0001+00 0.0002+00	2 6641-26		
0001+00 4 0411-07	0.0005400		
0001+00 2 0941-10	1 2441-14		
0001+00 2 2695-10	0.0001+00		
0001+00 1.0091-10	0 0001+00		
0001+000 0 0001+000	0.0001+00		
0001+00 1.7671-05	0.0001+00		
	0001+00 1 0311-10 0001+00 0 000+00 0001+00 3 107+05 0001+00 3 107+05 0001+00 1 1441-10 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 0001+00 0 0001+00 00	0001+00 1 031E-10 4 374E-11 0001+00 2 0000+00 0 0000+00 0001+00 3 187E-50 0000+00 0001+00 1 144E-10 0 0001+00 0001+00 1 044E-10 0 0001+00 0001+00 1 044E-10 0 0001+00 0001+00 1 044E-10 0 0001+00 0001+00 0 0001+00 0 0001+00 0001+00 0 0000+00 0 0001+00 0001+00 1 547E+05 0 0000+00 0001+00 1 547E+05 0 0000+00 0001+00 1 547E+05 0 0000+00 0001+00 0 0000+00 0 0000+00 0000+00 0 0000+00	0001+00 0 1001+00 1 001+00 0001+00 0 1001+00 0 0001+00 0001+00 1 141+10 0 0001+00 0001+00 1 141+10 0 0001+00 0001+00 0 0001+00 0 0001+00

Page 47 of 61

FIGURE A-2

	OUTPUT UNI		PAGE 24
	ATHFINDER ACTIVATION ANALYSIS. CURIES PER GRAM OF MATERIAL.		
POW	1. 0.000001-00 m BURNER 0.000001-00 mm 1104 5 001-13 MC 1104	ACTIVATION FR	DOUCTS
0	7 MACLIDE TABLE RADIOACTIVITY CLEUS		
	THE BOAL XSEC . FLUX AT CORE BOLNDARY . STAINLESS STEEL		
	STARTUP SHUTDOWN 1- JUN- 1989		
Y0170	0 0001 + 00 0 00+ 1000 0 00+ 1000 0 00+ 1000 0		
VB171	0.0001+00 0.0001+00 0.0001+00		
VB 172	0.000+1000.0 00+1000.0 00+1000.0		
YB173	0.0001+00 0.0001+00 0.0001+00		
¥8174	00+1000.0 00+1000.0 00+1000.0		
Y8175	0.0001+00 1.0441-04 0.0001+00		
YB 1754	0.0001+00 1.3391-04 0.0001+00		
YB 176	0 0001+00 0 0001+00 0 0001+00		
10177	0.0001+00 7 8431-06 0.0001+00		
11176	1 4171-14 7 3741-14 7 3741-14		
111768			
10177	0.0000 +00 1.1716 -04 1.4221 -23		
101774	0.000(+00 1.550(+07 6 143(+2)		
## 174	0 0001 +00 0 0001 +00 0 0001 +000		
HF 175	0.0001+00 3.0351+06 0.0001+00		
11 176	0.0001+00 0.0001+00 0.0001+00		
HF 177	0.0001+000.0.00+1000.0.00+1000.0		
+# 178	00+1000.0 00+1000.0 00+1000.0		
19 17 44	0 0001+00 1.6741-00 0.0001+00		
HF 179	0.0001+00 0.0001+00 0.0001+00		
HF 1796	0.000t+00 1.384t+04 0.000t+00		
HF 180	0.000+1000 0.00+1000 0.00+1000 0		
HFIBOM	0.0000+00 4.4941-07 0.0000+00		
HF 181	0.0001+00 3.0641+05 0.0001+00		
HF 182	6.000E+00 3.684E+15 3.684E+15		
14180	0.0001+00 0.0001+00 0.0001+00		
TA 181	0.0001+00.0.0001+00.0.0001+00		
14182	0.0001+00 1.4361-08 3 6841-15		
141824	0.0001+00 7.6451+11 0.0001+00		
	0.0000 +00 1.0671+04 0.0001+00		
	0.0001+00 0.0001+00 0.0000+00		
	0.0001+00 0.000(+00 0.0001+00		
	0.0000 +00 2.5246-04 0.0000+00		
	0.0000+00 5.0116-07 0.0000+00		
	0.0001+00 0.0001+00 0.000+000		
	0.0000 +00 8.7741+03 0.0000+00		
	0.0001+00 2.0101+06 0.0001+00		
	0.0001+00 5.0061-26 0.0001+00		
RE 185	0.0001+000.0001+000.0000+00		
RE :86	0 0001+00 5.9671+06 0.0001+00		
RE 187	0.0000+00 2.8481-14 2.8961-14		
REIAL	0.0001+00 2.4191-04 0.0001+00		
RE 1000	0.0001+00 2.3441-04 0.0001+00		
RE 189	0.0001+00 2.1331+09 0.0001+00		
05184	0.0001+00 0.0001+00 0.000(+00		
05185			
03186	0.0001+00 0.0001+00 0.0001+00		

Page 48 of 61

FIGURE A-2

PARTIAL ORIGEN2 SAMPLE OUTPUT

1	алел INIT		FAGI 25
	THE INDER ACTIVATION ANALYSIS. CURIES PER GRAD OF MATERIAL.		
PONT		ACTIVATION PEO	ACTS
120.00	THERE IS THE AT THE AT THE AT ANY AT AT AT AT		
	STARTLE SAUTTONE L. SAUTER		
05187	C DDDI +DC C DDDI +DC C DDDI		
05184	6 8001 -00 0 0001 -00 0 0001 -10		the set of the law bits
05185			
05190	0.0001+00 0.0001+00 0.0001+00		
05 1904			
05191	0 0001 000 1 3471 13 6 0001 00		
051918			
05162			
05193	0 0001+00 6 5 01-21 0 0001+00		
05194			
18101	0.0001+00 0.0001+00 0.0000		
18183	0.0001+00 0.0001+00 0.0001+00		
18164			The second s
181948	0 000/+00 3 004/+16 0 000/+00		
PTINO	0.0001+00 0.0001+00 0.0001+00		A STATE OF A STATE OF A STATE OF
	0 1001-00 0 0001-00 0 0001-00		
PT182	£ 0001+00 £ 0001+00 £ 0001+00		
P1183	0 0001-00 4 3411-23 4 4411-23		1 and Strail
PTINA	0 0001+00 2, 2101-10 0 0001+00		
FTIN	6 0001+00 6 0001+00 0 0001+00		
PT 195	0001+00 0 000+000 0 000		
PTION	0 0001+00 2 4741-23 0 0001+00		
PTINE	0.0001 +00 0.0001 +00 0.0001 +04		
PT197	5 1001 +L0 0 0001 +00 0 0001 +00		APPEND AND AND AND
PTYNTA			
P1198	0 0001+00 6 0001+00 0 0001+000		
PT 194	0.0001+00 0.0001+00 0.0001+00		
PTING	0.0001 +00 0.0001 +00 0.0001 +000		
ALITET	0.0001+00 0 0001+00 0 000+000		
	0.0001+00 0.0001+00 0.0001+00		and the state of the state of the
AU195	0 0001+00 0 0001+00 0 0001+000		
AU200	0.0001+00 0.000+000 0.0001+00		
-	0 0001+00 0 0001+00 0 0001+00		
HC197	0.0001+00 0.0001+00 0.0001+00		
	0.0001+00 0 00+1000 0 000+100		
HOINS	0.0001+00 t .000t+00 0 .000t+00		
	0.0001+00 0.0001+00 0.0001+00		
-	0.0001+00 0.0".at+00 0.0001+00		The second second
10200	0 0001+00 2 0001+00 0 0001+00		
HC201	6 0001+00 6 0001+00 6 0001+30		
160202	6 0001+00 6 0001+00 0 000+000		
HC203	0.004 .+00 0.0001+00 0.0001+00		
HC704	0 0001 +00 0 0001 +00 0 0001 +000		
10205	0 0001 +00 0 0001 +00 0 0001 +50		
11 203	0.0001+00 0 0001+00 0 0001+00		
TL 204	0 0001+00 0 0001+00 0 0001+00		다 도망 가지 못하는 것이 좋아.
71.205	0 0001+30 0 0001+00 0 0001+00		
1.204	0 0001+00 2 6091+14 4 6101+21		
P8204	1.15(1-20 1.1561-20 1.1561-20		
P8.205	0 0001+00 + 3491+14 1 3491+14		
			And a first state of the second state

Page 49 of 61

FIGURE A-2

• PATH POREE • 0206 0207 0208 0208 1209 1209 1209 1209 1210 0210 0211 0210 0211 0211 0210 0211 0211 0218 POREE •	INDEE ACTIVATION ANAL 0.000001+00 mm PLON 7 NUCLION STATUS Selficter 0.0001+00 0.001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.0001+00 0.4271+10 0.0001+00 0.0000+00 0 0.2511+12 5.5551+00 2	LV515. CUR11 NLF+ E GODODI TABLE EADIO EBAAL KSIC - 1-JUA-1984 0 ODDI+00 0 ODDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 DDDI+00 0 0001+00 0 0001+00 0 0001+00 2 4191-03 0 DDDI+00 1 4191-03 0 DDDI+00 0 DDI+00 0 DD	15 PER CALL	L OF BATTETA FLUXA 5 .001 CURIES HE DOLADARY	L. +13 N/GH++2-SEC + STAINLESS STI	R.	ACTIV- ISON PRO	XR.CTS
PCREE+	C DODODI-DD C C C C C C C C C C C C C C C C C C	N.F. C DODDD TABLE EADIO (BAAL ESEC - 1-JLN-1640 0.0001-00	HOD MED.	FLURE S DOF CURIES WE DOLADARY	+13 N/G#+12-SEC + STAINLESS STO	L		*****
206 207 208 206 206 206 210 210 211 210 211 211 211 211 211 211	7 NUCL (DE 1 THE STATUS SelfIces 1 0 OD01+00 0 OD01+00 0 0 OD01+00 0 AD01+00 0 2511-12 5 5551+00 2 0 0001+00 0 000(+00 0	TABLE BADIO BRAAL KSIC - 1-JUA-1984 0 ODDI-00 0 ODDI-00 0 DDDI-00 0 DDI-00 0 DDI-00	ACTIVE TAB	CURIES HE BOLMENEY	+ STAINLESS STI	R.		
1204 1207 1208 1208 1208 1208 1208 1208 1210 1210	THE 514114 Set7100000 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 4 201+13 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 0 0001+00 4 4201+14 1 2511+12 5 5551+00 2 2511+12 5 5551+00	EMAL KSIC - - JUN-1989 0 0001-00 0 0001-00 2 0191-00 2 0191-00 2 0191-00 0 0001-00 1 0000-00 1 0000-00 1 0000-00 1 0000-00 1 0000-00 1 0000-0	FLUR AT CO	RE BOLADUEV	+ STAINLESS STI	8.		
206 207 208 206 206 206 205 210 210 211 211 211 211 211 211 211 211	STARTUP SPECTRAMY 1 0 DD01+00 0 0 DD01+00 0 0 DD01+00 0 4 2721+13 0 0 D001+00 0 4 000 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	1 - J.N. 1989 0 - DODI - DO 0 - DODI - DODI - DO 0 - DODI - DODI - DO 0 - DODI	LATIVE TAB	LE TOTALS				
206 (207 (208 (208 (206 (206 (206 (206 (207) (20	6 0001+00 6 0001+00 1 6 0001+00 6 0001+00 1 6 0001+00 6 0001+00 0 6 0001+00 6 0001+00 0 6 0001+00 6 0001+00 0 0 0001+00 6 40271+13 0 0 0001+00 6 40271+13 0 0 0001+00 8 4021+13 0 0 0001+00 8 4021+14 0 0 0001+00 3 6481+20 0 1 2511+12 5 5551+00 2 0 0001+00 6 000(+00 0 1 2511+12 5 5551+00 2	C 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 4441-21 0 0001-00 0 4441-21 0 0001-00 0 4441-23 0 0001-00 2 4191-03 0 0001-00 1 4191-03	LATIVE TAB	LE TOTALS				
(907 208 206 206 206 206 210 210 210 211 210 211 211 41 7AL	6 0001+00 0 0001+00 0 0 0001+00 0 0001+00 0 0 0001+00 0 7 181-08 0 0 0001+00 0 0001+00 0 0 0001+00 0 0001+00 0 0 0001+00 0 4 0001+00 0 0 0001+00 0 4 0001+00 0 0 0001+00 0 4 0001+00 0 1 2511+12 5 5551+00 2 0 0001+00 0 000(+00 0 1 2511+12 5 5551+00 2 0 0001+00 0 000(+00 0	0 0001+00 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 2 4191+03 CumJ 1 4191+03	LATIVE TAB	LE TOTALS				
2004 2004 2004 2004 2009 2100 2110 2110 2111 2211 22110 22114 714L ************************************	C 0001+00 C 0001+00 C C 0001+00 C 0001+00 C C 0001+00 C 0001+00 C C 0001+00 C 0001+00 C C 0001+00 C 4272+13 C C 0001+00 C 4272+13 C C 0001+00 C 4271+13 C C 0001+00 C 422+15 C C 0001+00 C 422+15 C C 0001+00 C 024+14 C C 0001+00 C 00C(+00 C C 2511+12 S 5551+00 C C 0001+00 C 00C(+00 C C 000(+00 C 00C(+00 C 00C(+00 C C 000(+00 C 00C(+00 C 00C(+00 C C 000(+00 C 00C(+00 C 00C(+00 C 0 C 00C(+00 C 0))	C 0001+00 D 00000+00 D 0000+00 D 0000+000 D 0000+00 D 0000+00 D 0000+00 D 0000+00 D 0000+00 D 0000+00	LATIVE TAB	LE TOTALS				
204 204 209 210 210 211 210 211 211 211 211 211 211	0 0001-00 0 0001-00 0 0 0001-00 0 0001-00 0 0 0001-00 0 0001-00 0 0 0001-00 0 4 8271-13 0 0 0001-00 0 4 8271-13 0 0 0001-00 0 4 801-21 0 0 0001-00 0 4 801-21 0 0 0001-00 0 4 801-14 3 0 0001-00 0 8 801-20 0 1 2511-12 5 5551-00 2 0 0001-00 0 0000 -00 0 1 2511-12 5 5551-00 2	0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 0 0001-00 2 4191-03 0 0001-00 1 4191-03 0 0001-00 1 4191-03	LATIVE TAB	LE TOTALS				
2005 210 210 211 210 211 2110 2110 2110	0 0001+00 0 0001+00 0 0 0001+00 4 4271-13 0 0 0001+00 4 4271-13 0 0 0001+00 4 4271-14 0 0 0001+00 4 4301-14 0 0 0001+00 3 4301-14 0 0 0001+00 3 6381+20 0 1 2511-12 5 5551+00 2 0 0001+00 0 0000+00 0	0 0001-00 0 0001-00	LATIVE TAB	LE TOTALS				
210 210 211 210 210 211 210 211 210 74 74 74 74 74 74	0 0001+00 4 4271+13 0 0 0001+00 4 4501+21 4 0 0001+00 4 4201+14 0 0 0001+00 4 4201+14 0 0 0001+00 3 1641+14 0 0 0001+00 3 1641+20 0 1 2511+12 5 5551+00 2 0 0001+00 0 00001+00 0 1 2511+12 5 5551+00 2	0 0001 00 0 0000 00 0 00000 00 0 0000 0000 00 0 0000 00000000	LATIVE TAB	LE TOTALS				
2160 211 2210 2210 2210 2210 2210 2210 2	0 0001+00 4 8501-21 4 0 0001+00 4 4501-14 5 0 0001+00 3 4501-14 5 0 0001+00 3 4541-20 0 1 2511-12 5 5551+00 2 2 2511-12 5 5551+00 2 0 0001+00 0 000(1+00 0 1 2511-12 5 5551+00 2	4441-21 0 0001-00 0 4601-23 0 0001-00 0 6001-00 0 6001-00 0 6001-00 2 6191-03 0 0001-00 1 6191-03 0 0001-00 1 6191-03	LATIVE TAB	LE TOTALS				
211 2210 2211 2211 2211 2211 2211 2211	0 0001+00 4 1421-14 0 0 0001+00 4 4301-14 3 0 0001+00 3 1441-14 0 0 0001+00 3 1441-14 0 1 2511-12 5 5551+00 2 0 0001+00 0 000(1+00 0 1 2511-12 5 5551+00 2	0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00 0 0001+00	LATIVE TAB	nt 101413				
1210 1211 1211 14 14 14 14 14 14 14 14 14 14 14 14 1	0 0001+00 # 4101+14 3 0 0001+00 3 6441+14 0 0 0001+00 3 6441+14 0 1 2511+12 5 5551+00 2 2 2511+12 5 5551+00 2 0 0001+00 0 000(1+00 0 1 2511+12 5 5551+00 2	0 4401-23 0 0001-00 0 0001-00 2 6191-03 CumJ 0 6191-03 0 0001-00 1 6191-03	LATIVE TAB	nt 101413				
1211 12114 1741 *FF 1xFP *AC1+FP	0.0001+00 3.0381+20 0 0.0001+00 3.0381+20 0 1.2511+12 5.5551+00 2 0.0001+00 0.000(+00 0 1.2511+12 5.5551+00 2 0.0001+00 0.000(+00 0 1.2511+12 5.5551+00 2	0.0001+00 0.0001+00 2.0191-03 CumJ 0.0191-03 0.0001+00 1.0191-03		nt 101413				
12114 1741 	0 0001+00 3 0381+20 0 1 2511-12 5 5551+00 2 2 2516-12 5 5551+00 2 0 0001+00 0 000(+00 0 1 2511-12 5 5551+00 2	0.0001+00 2.6191-03 CumU 2.6191-03 0.0001+00 1.6191-03		LE TOTALS				
141 1479 1479	1 2511-12 5 5551+00 2 2 2511-12 5 5551+00 2 0 0001+00 0 000(1+00 0 1 2511-12 5 5551+00 2	Cumu Cumu 2 6191-03 0 0001-00 1 6191-03		LE TOTALS				
+## 1x## +AC1+##	2511-12 5 5551+00 2 0.0001+00 0 000(1-00 0 1.2511-12 5 5551+00 2	0001+00 1 6191-03	LATIVE TAB	LE TOTALS				
486 1x89 •AC1+89	2 2516-12 5 5556+00 2 0 0001+00 0 0003+00 0 1 2516-12 5 5551+00 2	0 0001+00 0 0001+00 0 6 192-03						
1+FP +AC1+FP	00001+00 0 00001+00 0 1 2511-12 5 5551+00 2	0001+00 6191-03						
*401+88	2511-12 5 5551+00 2	6191-03						

Page 50 of 61

APPENDIX B

PATHFINDER CURIE ESTIMATE SPREADSHEET

Page 51 of 61

APPENDIX B

PATHFINDER CURIE ESTIMATE SPREADSHEET

Using the curie per gram source terms for 1 June 1989 calculated for the various materials as typically shown in Appendix A, a Lotus 1-2-3 spreadsheet was constructed to perform the disparate tasks necessary to determine the total curie contents of the vessel package and individual components as of 1 January 1990. These tasks are:

- Further decay of the curie/gram source terms to 1 January 1990 (the spreadsheet shown has already decayed these values from 1 June 1989 to 1 January 1990; therefore the decay factor for all radionuclides is shown as 1.00
- Calculation of logarithmic averages for various regions and constructing a set of radionuclides appropriate for this average
- Determination of individual weights of materials in each region of the neutron activation model
- Calculation of curie contents by isotope in each component in each region of the activation model
- Summation of curie contents by isotope of each component
- Summation of curie contents by isotope for entire vessel package
- Summation of curie contents for entire package and determination of relative contribution by isotope

The regions referred to in the spreadsheet are those shown in the neutron activation model as described in Section 5 and depicted in Figures 5.3 and 5.4. Region numbering begins with the central cylinder, uppermost layer. Region numbers increase with decreasing elevation until the bottom of the central cylinder (region 7), then begins again with region 8 at the top of the first concentric ring around the core.



SECTION FIGURE PATHFINDER CURIE EST UPPER LEFT

-

ALTERNAL DENSITIES (CAUCHTE):	COMMAN	-	-					-	NIRON CRO	5-51210	COMMECTS	ON FACTORS			
Stainless steel SFR. 7.90 Zircaloy-2 0.57 Carbor steel A312 3.09		6 8 2 2 1 8 8		tione Core		10-1000 10-1000			The second secon	anter a contra	88888 19191	:: : n:::	*****		
trout lisotopic decay date. 01-	**			etes core		CO-1000									
Cut cutinder getchegrine attes v		2	3	*		5	Control	5	-	1	8.5			;	•
0 3 501-13 Core region stainless	steel yout	2.581-07	80-168 .	54-391 C	16-14	1 201-02 1	1 10-161	10-100	20-322-62	3 .21-15	5. 361-17	80		-	- 190
0 3 SOLUTS CORE region Zircalov-		2 198-13	1 746-06	0 00-100 0	90-100	3.231-00 8	10-102		3.415-07	12-304 8	£4-394 B		0.00 2		
3 Sofert3 Core budry 55 3041 (b	witter shroud?	3 485-07	1 375-07	1.728-15 1	11-198	2 191-05 1		10-358.1	10-140 1	\$1-362 ·	1 - 129 - 18		0.00 7	301-10	11-300
1 SOLATE Core budry 55 3041 (e	scent throad	4 12E-08	+ 621-58	2 041-14 2	218-12	1 90-109 2	1 50-30+	1021-07	1 271-05	81-310.2	11-160 1	8.0			24-342
1 1 875+12 - 100 average for r		* *** *	3 11E-00	5. 901-16 4	291-15	2 10-306	2 10-30+	031-00	2.445-96	11-345 .	00-100 0	91-191.0	8 -346-18	11-199	21-145
2 3 301+10 Wess! Clad staintess	I STeel 3041	10-308 +	01-105 S	1.716-15 E		5 80-315 B	E 10-101		4.705-07	11-100 L	12-1m1 -1	8.0		201-12	C1-190
3 305-10 inner vessel mil car	thon steel A212	2. 151-04			134-34	1. 335-97 4	2 00-36.0		3. 101-00	. 021-16	22-305 .			* 61-36L	
3 9 081-09 - 100 average for 1		1.245-10	3 768-11	€ 51-34€ .	\$31-14	* #0-35# #	S 80-385		1.651-04			11-14 9	* 43-360 *	2 Et-102	235- 14
2 501-00 Outer wessel mil car	then steel A212	3.716-12	11-342-1	1.031-14	131-161	1.776-04 5	2 40-165		10-145 5		0 100 . 0	2.191-17	1 - 345 - 1	746-14 7	166
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2 1.01041 è 5 were that the input Cirgs

tegion contents	concrete balle beller control baldes control rod guide tubes control rod guide tubes	Superioration fuil cleaning takes superiorate fuil cleaning belier control blades forming red guide takes superiorate control radio	Superiorates fuel element tubes a cleaned builter bound Single element builter bound superiorates builter builter control broken	beller element bele den streeture Sterenbaster contra con Streethaster felle inter- streethaster belle inter- tieten belle beset	superintator ballite Instruction to sample follows Superintator (sub steamed tabes doiler control blacks
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der Cytinder description	o Core region				
tepion Cylin	•	•			

Page 52 of 61

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PATHFINDER CURIE ESTIMATE SPREADSHEET LOWER LEFT SECTION

-					1				al supports							*******						
Region Co		Anties or is a		oller baltie	offer balfie	ofter befile	citer balitie	Tene separat			esses classic	Prest cladely	ipees same	essel cleddi	Tiber Clade							
-	1	CO-100		6 CO- 100							A 60-100		001-100			A 60-300	A 20-106		A 10-100	A 20- 100	-	
		-		25 304	-	-	-						-			0 220	2101 2		C. 4313	222 22		
-	8. 5.	723.730	****										122 223	21		247.472	247.472		247. 172	20. 02		
	1		-										-						121.44	-		
	;						:												-			
taves description	s sat layer below core	. 2nd layer below core	" 3rd 13,er Leton core	316 layer above core	I 2nd fayer above core	I IST Tayer above core	· Reactor Core Level	t tel laver below core		316 layer belos core	. 3rd faver above core	The layer above core	Bear tor Core Level	tet laver belde core	Jed layer belos core	3rd layer above core	2nd layer above core	Beartor Core Level	tes taves before core	The fayer beine core		
-					-														•	• •		
											187.94											
Cytinder description				Steam separator region							vesset cladding region					Vessel mil region						TOTAL FOR VISSEL PACANON
trajor to Officier	•	•	•	•	•		:	P				**		::		2 2			*	. 1		

.. Soller baille above and below core does not have a flux red ction factor due to the use of homogenized Civins source term

SUMMERY OF CALES BY CONTOUNT	Tot motel		8	3	:	8	I	8	5	1	1			:
Superheater baltie	1. 306. 12	=	1. 071-02	20-100		M-140 6		10-100 6		10-104 C				
Superheater fuel Insul, tubes (Immus & outer)	7. 592.7	:	1.516-01	20-301 5		7 341-06	776-00		10-165 6	1 218-01	2315-00			
Superheater support plate			30-195 5	1 361-96	4. 356- 14	1 796- 10 2	975-94	1 501-03	30-306		5 361 - 14		791-12 3	636-13
Superheater control rods	25. 515	25.638	7.411-04	1.821-04	·	31-00 ·	20-315	10-102 6	1 745-03	10-340 2	1. 108-11	E1-105 1		
Boller fuel boxes	7.664.1	100.001	1 335-03	20-345 6	01-109-6		10-356	1. 191-00	20-299 5	- 356		10-112	1 10-101	291-01
Boller shroud	1. 050.00	100 001	1.926-01	20-100 5	7.546-10	8 161-DE 0		10-291 6		10-109 9		13-306 1		
Boller elemnt hold down structure	1 352 30		1 00-150 1	60-100 G	1. 785-10	S. 10- 101 .	10-100	. 58-185 .	20-198 2			C1-101 0	1 125-04 7	
Bolter Of guide tubes & remining structure	2.942 83	61.368	8 431-03	216-03	1.031-10	4 10-1+1 P	10-116	00-149 E	10-100		1 211-10	1 748-13		
Soller element polson shims	376.00	100 001	3-345-67	29-362	01-15a 6							E4 14.6 1	C -11- 04 2	
Boller control blades	1.816.1	84 278	1. 101-01 1	10-101		·	341-00		10-305 0			11.10		
Instrumentation tubing & samle holders	SO	848 604	C 101-101 0	10-161	1 001-100 1	A 18-310. 4	746-01	3 501-00			. 345-16	· ·····	725.00	
Boller grid plate	5.400 00	100 001	4 345-03	29-169	S 301-10	2 131-161 2	526-00			1 218-01		CA-180 0	1 34.00	
Steam separators & supports			1 301-03	20-30P		6 10-319 9	10-164		231-02		ISO I			BME - 14
feedwater ring & supports	1.027.00	100 001	2 651-03	10-165	01-120 L	19-326 1	10-165	. 201-01	1 266-03	1.375				-36- 10
Verse! cladding	4 235 25		1 196-93		1 131-04	9 921-00 6	341-07	10-100 6	10-119 1	19-121 1	1 338-14		2 44 - 142	
were last	32 005 44	34. 27	1.061-03	121-64	7. 161-00	1 754-67 4	10-161	1 336-91	- 335-94	24-110 .			6 01-18-	
Totals	18. 200. 18	***	29.0				* *	394 64						
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PATHFINDER CURIE ESTIMATE SPREADSHEET UPPER CENTER SECTION

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		1.081+03	:	17741 45 25	331-05		61-100 L		19-111 1	1 821-83					
		3.521-03	:	96322 96 7	50- 109	2 846 -05	81-14C 4	1 736-04		1 135-03	1 795.64				
						1 645 .04									
			-						-						
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		3. 526-03		1 36 225 36 7	40- DA	10-100 E	20-12	1 7 16 -00	CO- 520 0	1 171-01					
		1 605-01		4 07 076F	675-05	2 341-05	E4-381 1	- 322 -	10-100 1	20-210 2					
		10-365 E	-		801-04	50-J00 9	CA-160 1		50-102 .		344-94				
		1.041-03	*	20177 06 3	346-03	1 275-03	11-100	10-109							
			32	14774 50 5	20-199	. 311-04	1.036-111	131-00	CO-103 .	1 251-01					
		8 171-03	•	6174.18	336-04	3. 171-04	1.021-11	- 100	4 851-02	10-109 6		10-111 0	1 236-51		
		1 081-03	:	17241.43 2	336-03		1. 11-11-1	10-101 I	19-156 1	1 025-000	10-115 5				
		3.321-03		56322 56 7	60-109	2. 84C-03	1 205-11	1.736-07		3 325+00	20-382 1	21121-00	121 - 10		
		7.765-04		775.44 94 94 1	28-260	50-344 E	01 - Jac 1	10-101	10-100 9	4 585-06	20-190 2				
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		10-365.2			80-109	. 001-04	1 11-354-1	80-368	20-342 0	10-105 9	E0-192 E	10-398 8	3 345-11	81 325 6	
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Page 54 of 61





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			10-165 2	1	11032 24	1 601-04		- 21 - 156 · I	10- He'	60-jet .		-02 3 76		BE-02 2	1 366-12	3.326-5	1.701-1
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			255 221	•	122.552	1. 151-03	3.431-94	5 40-166	321-06	1. 121-02	3.325	01 2 529		1 10-316	11-181	7.321-1	1 701 - 1
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			241.112	•	241 195					576-03	-			2 60-16	51-306	- 100 a	1-119.8
					001 000	CO-170 .			10-311	10-1-0		12 1 10		20-101	11-125		1-362 5
										CA-216				2 64-16	CI - 106		
			247. 172	••	241 172	10-100 1		6 11-161	11-195	10-11C				2 NO-14			
ATTR	FOR VISSEL PACKACE					10-352 9	7 10-319.5	C 10-344	50-3m	10-101	3 576.	62 1.50E	- 8-	. 20-10	766-90	-14P 1	0-101 I
191	pic percentage of total					0.138			0 00	7.058	55	-	316	37.538		8 0	

Page 55 of 61

SIMMARY OF CIDITS BY CONTINUES									
				CF105		1010	FCIAL		
Superheater halfle	3 071-04	1 248-03	0 001-00		741-67				
Superheater fuel incut tides those & auters									
					- 1CF - C	FL-160 .	10-111-	CA-180 C	
Supermenter support plate	1.791-08	00-309 S	0.001-000	11-120.5	1. 245-11	4. 378-17	1. 131-06	1.271-07	00.0
Superfreater control rods		1. 161-05	00-300 0	10-1+0" I	3 501-00	SI-101 .	3.351-04	2.641-05	0.9
Boller fuel boxes	1. 355-05	2.115-04	3.541-02	14-107 9	1 301-01	E1-169.1	6 366-03	4. BOE-DA	19.97
Boller shroud	3.795-04	30-390 5	00-100 0	1.325-06	3 276-07	11-316.1	1. 451-03	2.665-03	29 BOI
Soller elearnt hold down structure	30-350-62	. 651-94	0.001-00	10-388.1	3.451-00	1. 241-13	3 326-03	3 735-04	
Boller OR guide tubes & remining structure	\$0-10F 9	1.331-04	00-300 0	10-341 1	2 941-04	1 Dat - 13	2 675-03	3.012-04	15.9
Boller elemnt polson shins	2.451-04	5. DBT-64	00-100 0	10-195 .	19-161 1		1 671-02	1 151-03	25. 54
Soller control blades	* 335-04	1 736-03	00-300 0	1 555 06	3 821-07	1 356-12	20-388 5	1 474-65	
instrumentation tubing & sample holders	\$0-312 .	10-362 1	00-100 0	1. 151-07		E1-100 .	50-105 2	7 929-04	
Bolter grid plate	30-362 6		00-100 0	10-311 9	19-116 1	51-155 5	1 376-02	60-365 I	33.76
Sieau separators & supports	1.011-05	1.746-06	0 001-00		a 766-90	00-100 0	10-121 1	1 616-04	
feeduater ring & supports	3. 121-06	10-104 .		01-Jas 6	- 346 · 10	00-100 0	3. 175-94	- 535 - 04	. 7.
Vessel cladding	3. 121-06	1.031-07	00-100 0	51-181 5	E1-168.1		3 401-De	2 001-05	6.72
	1. 441-06	10-316-1	0 001-00	61 - 326 - 83	£1-309 1		2 341-03	1 341-54	
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PATHFINDER CURIE ESTIMATE SPREADSHEET UPPER RIGHT SECTION

Document N04-22-005

Page 56 of 61

10-10-10-10-10-10-10-10-10-10-10-10-10-1
10-16-17 0 1-12(-02 10-16-17) 0 391-01 10-16-17 0 391-01
10-165 0 10-166 1 20
10-125 2 50-151 1 20




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						50-112	50-115-2	10-100 D	3 256 - 61		-00	1-396		-0-31	50-349 5	1. 346-00	0.761
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						50-320	10-3+1-61	0.001-00	10-100 1	3.67		0-100			20-310-1	3.471-00	0.739
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5								00-100 Q	2.501-14			0-100			50-380	10-100 .	6.1.0
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					£ 61-307 S	101-04	01-320 0	00-100 0	1 041 - 14	3.60	- 13 0	10-100				1.625-63	
					C1-160 S	10-101	0-120 G	00-100 0	1.0et-13	3 60	0	0-100		2 50-1	90-101	10-120-1	0.00
*					£ 21-360 S		10-120 L	0.100 0				- 100	2	21		10-122 .	
							01-10	0.001-00	1 - Jao - L	2 601			0.0	- 8-1	10-141	1 026-03	0.00
						01-101	11-120.0	00-100 0	1.041-15	3.60		- 100	3.74		- 141	1.025-04	0.00
	101	THE FOR WES	SEL PACAGE		- 231-03 C	60-150	60-JOL .	3. 541-02	10-10F 9			-14		1.	20-160	10-194 .	
						5.00X	0.00		0.13								

Page 57 of 61

Page 58 of 61

APPENDIX C

MICROSHIELD INPUT AND OUTPUT DESCRIPTION

Page 59 of 61

APPENDIX C

MICROSHIELD INPUT AND OUTPUT DESCRIPTION

MICROSHIELD Version 3.12 was used by TLG Engineering to perform the calculations required to estimate the shielding thickness for packaging the Pathfinder reactor vessel and internals. The computer code was verified and validated in-house using benchmark problems from ANSI 6.6." "Calculation and Measurement of Direct and Scattered Gamma Radiation from LWR Nuclear Power Plants".

An extensive number of calculations were made to determine the exposure rate two meters from the surface of the shipping package. For the purposes of this analysis, the vessel and its internals (the source region) were divided into four concentric regions and three axial regions as follows:

- A. **Concentric Regions**
 - 1. Core region.
 - 2. Steam separator region.
 - Vessel clad region.
 - 4. Vessel wall region.
- B. Axial Regions
 - 1. Core height.
 - 2. One foot segment above the core.
 - One foot segment below the core.

Exposure rate contributions were calculated for each of the twelve regions for various package shielding thicknesses and packing densities of gravel (concrete) in the vessel. Additionally, a calculation was made for various packing densities of gravel at the core centerline, one centimeter from the reactor vessel outer wall. This calculation was made in order to benchmark the shielding calculations to exposure rate measurements made at Pathfinder by the plant staff.

Figure C-1 is a MICROSHIELD calculation (input and output) for the exposure rate at two meters from the package surface due to the core region internals, for the height of the core, with twenty-five void percent gravel and a two inch thick carbon steel shield. The source term for the region, as for all regions, was taken from the curie spreadsheet as shown in Appendix B.

Page 60 of 61

FIGURE C-1 MICROSHIELD SAMPLE PROBLEM

			Nicroshi	eld 3.12		
		(TLG Enginee	ring . #141)	
Page :	1				File Ref:	
File :	RXSHD101.	MSH			Dete:	_1_1_
Run date:	October 3	, 1989			By:	
Run time:	9:07 a.m.				Checked:	
•	ASE: Core	Region · Co	ore Height -	25 v/0 cor	wrete · 2.0	in. shid.
	GEOMETRY	7: Cylinde	r source fr	om side - d	cylindrical	shields
istance	to detector			x	403.2	cn.
Source le	ngth				182.880	
Dose poin	t height f	on base		Y	91.440	
source cy	Linder rad	ium		11	87.630	
hickness	of second	shield		12	79.375	
Thickness	of third a	shield		13	0.640	
Ih ickness	of fourth	shield		14	7.620	
Inickness	of fifth a	shield		15	5.080	
licroshie	ld inserted	d air gap		sir	222.855	•
		Source Volu	me: 4.41186	e+6 cubic c	entimeters	
		-	TERIAL DENS	TIES (g/cc):	
Asterial	Source	Shield 2	Shield 1	Shield 4	Shield 5	Air pap
	•••••		••••••	•••••		
luminum						.001220
arbon						
oncrete	1.5390	1.7280				
Iron	.32620	. 15560	7.90	7.890	7.90	
B. B. C.						
COU						
ithium						
ithium						
ithium ickel in						
ithium lickel in itanium						
ithium lickel in itanium ungsten						
ithium lickel in litanium ungsten Irania						
ithium lickel in litanium ungsten Jrania Iranium						
ithium Nickel In Itanium Ungsten Irania Iranium Later						
Lithium Rickel Iin Titanium Ungsten Irania Irania Iranium Iater Tirconium	.5620					
Lithium Rickel Tin Titanium Ungsten Jranium Jater Tirconium	.5620					
ithium (ickel in (itanium Ungsten Ungs	.5620					

Page 61 of 61

FIGURE C-1 MICROSHIELD SAMPLE PROBLEM (continued)

Page 2 File: RXSH101.MSH CASE: Core Region - Core Neight - 25 v/O concrete - 2.0 in. shid

> BUILDUP FACTOR: based on GP method Using the characteristics of the materials in shield 5.

INTEGRATION PARAMETERS:

NUMER	er of	lateral angle a	egments (Nthets)	15
Numb	er of	azimuthal angle	segments (Npsi)	15
Numb	er of	radial segments	(Nradius)	15

SOURCE NUCLIDES:

Co-60: 1.6390+12 curies

RESULTS:

Group	Energy (HeV)	Activity (photons/sec)	Dose point flux MeV/(sq cm)/sec	Dose rate (mr/hr)	

1	1.3359	6.064e+12	2.407+00	4.344e-03	
2	1.1797	6.064e+12	9.331e-01	1.7340-03	
3	.6953	9.892+08	1.815e-06	3.7370-09	
4					
5					
6					
7					
10					
11					
12					
13					
14					
15					
16					
17					
18					
19					
20					
1	OTALS:	1.213+13	3.341++00	6.078e-03	

APPENDIX 5.5.2

References

- 5.1 ORNL/TM-7175, "A user's Manual for the ORIGEN2 computer Code", Croff, A.G., Oak Ridge National Laboratory, July 1980
- 5.2 Microshield 3 Reference Manual, Software Version 3.12, Grove Engineering, Inc. Rockville, MD, 1988



TABLE 5.1 SUMMARY OF MAXIMUM EXPOSURE RATES (mR/hr)

NORMAL CONDITIONS OF TRANSPORT: EXCLUSIVE USE VEHICLE

	Pac	kage Sur	face*	2 M Surface	2 Meters From Surface of Package		
	Side	Top	Bottom	Side	Top	Bottom	
Gamma	38	Negl.	Negl.	6.9	Negl.	Negl.	
Neutron	N/A	N/A	N/A	N/A	N/A	N/A	
Total	38	Negl.	Negl.	6.9	Negl.	Negl.	
10 CFR 71 Limit	200	200	200	10	10	10	

.

Package surface top refers to the forward end on the railcar, and package surface bottom refers to the aft end. Rev. 0

6.0 CRITICALITY EVALUATION Not applicable. 7.0 OPERATING PROCEDURES

Not applicable.

8.0 ACCEPTANCE TESTS AND MAINTENANCE PROGRAM

The Pathfinder package is designed to be a one time use package for the disposal of the reactor vessel and internals. This chapter describes the acceptance tests required to place the package on a railcar for transport. No long term maintenance program is required since this is a one time shipment.

8.1 Acceptance Tests

Acceptance tests for the Fathfinder package will include tests for shielding and containment integrity, and a visual inspection of each stage of the package preparation. Pressure tests and leak tests are not required.

8.1.1 Visual Inspection

Prior to the attachment of the Hexcel impact limiter material, the vessel exterior surface will be visually checked for nozzle cover plate weld integrity, cylindrical shield weld integrity, head closure bolt damage. After attachment of the Hexcel material but before cover shell welding, a visual inspection of the Hexcel condition will be performed. Any damage to the Hexcel material will be evaluated as to its importance to the impact limiter performance. If deemed unacceptable, that portion of the Hexcel will be repaired or replaced as necessary. Upon application of the Hexcel cover shell and final welding, a visual inspection will be made of the cover condition.

If the inspection reveals any defects they will be evaluated as to their significance to the shipping package. Any modifications to the package design will be reviewed before proceeding, if required. All inspections and repairs will be appropriately documented with Quality Control inspection reports.

Prior to leaving the Pathfinder site, an inspection of the package will be made to verify that it was prepared in accordance with all requirements. The overall package will be visually inspected for any defect or unusual condition.

8.1.2 Structural and Pressure Tests

The package contains no removable closures, and no pressure tests are required.

Structural tests will include nondestructive tests of new welds and closure integrity including the shield to vessel weld. Nondestructive tests will include liquid penetrant or magnetic particle examination weld integrity inspections. These inspections will be made Rev. 0



8.1.3 Leak Tests

The package contains no liquids or contained gases and leak tests are not required.

8.1.4 Component Tests

The package is of passive design and contains no ancillary components which would require testing.

8.1.5 Tests for Shielding Integrity

Radiation and contamination surveys will be performed prior to shipment of the package to ensure the external dose rate and contamination levels are in accordance with 10 CFR 71. The acceptance limits are:

- 1)200 mRem/hr at any point on the vertical planes projected from the outer edges of the package surface, on the upper surface of the load.
- 2)10 mRem/hr at any point 2 meters from the vertical planes projected from the main edges of the package.

8.1.6 Thermal Acceptance Tests

The package will not be affected by the small thermal load produced by the contents (4.67 watts) and there are no thermal acceptance tests.

8.2 Maintenance Program

This is a one time use container and no maintenance program is required.

APPENDIX 8.3

References

8.1 American Welding Society Test Procedure for Nondestructive Weld Test Methods.