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LOFT STEADY STATE CRITICAL HEAT FLUX TESTS (6.9 to 13.8 MPa)

RICHARD C. GOTTULA

June 1978



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LOFT STEADY STATE CRITICAL HEAT FLUX TESTS (6.9 TO 13.8 MPa)

by

Richard C. Gottula

EG&G Idaho, Inc.

June 1978

PREPARED FOR THE U.S. NUCLEAR REGULATORY COMMISSION AND DEPARTMENT OF ENERGY IDAHO OPERATIONS OFFICE UNDER CONTRACT NO. EY-76-C-07-1570

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ABSTRACT

Steady state critical heat flux (CHF) tests have been performed on electrically heated rod bundles simulating the central region of the Lossof-Fluid Test (LOFT) nuclear reactor core. Previously reported steady state CHF tests have shown that cladding surface thermocouples on LOFT fuel rods reduce the critical heat flux over the pressure range of 13.8 to 16.5 MPa. Additional steady state CHF tests, reported herein, have been performed to determine the effects of rod external thermocouples on CHF over the ranges of pressure (about 11 MPa) and quality (30 to 40 percent) where CHF is predicted to occur in LOFT during blowdown operation and to determine if sufficient data could be obtained to develop a CHF correlation to predict critical heat fluxes in the LOFT core for blowdown operation.

These low pressure tests were conducted on similar electrically heated 25-rod bundles, which simulate the central region of the LOFT core, both with and without rod external thermocouple simulators. The test data indicated that rod external thermocouples do not significantly affect the magnitude of CHF in the pressure and quality range where CHF is predicted to occur in LOFT during blowdown operation. Therefore, based on these results cladding surface thermocouples are not expected to have a significant effect on transient critical heat flux or time-to-CHF for the nuclear rods in the LOFT core under blowdown conditions. These conclusions are consistant with previously reported transient (blowdown) CHF tests which have shown the same results.

From these initial tests, an insufficient data base was obtained over the pressure range of 6.9 to 13.8 MPa to develop a new correlation that could potentially be used to predict critical heat fluxes for LOFT under blowdown conditions. This resulted from experimental facility limitations on fluid quality during these tests.

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SUMMARY

A steady state experimental program was performed to investigate the behavior of critical heat flux (CHF) in electrically heated rod bundles simulating a portion of the central assembly of the Loss-of-Fluid Test (LOFT) nuclear Core-1. The effects on CHF due to cladding surface thermocouples attached to some of the LOFT fuel rods were investigated over the pressure range where CHF is expected to occur during a loss-of-coolant-experiment in the LOFT facility. Twenty five-rod bundles, both with and without simulated cladding surface thermocouples, were tested.

The ranges of the experimental parameters investigated were: pressure: 6.9 to 13.8 MPa mass velocity: 680 to 3400 Kg/s-m² local quality at CFF: -0.25 to 0.60

One-hundred-fifty-seven data points were obtained with the rod bundles having rod external thermocouple simulators. These data were compared with 53 data points obtained with the rod bundles without external thermocouple simulators. The comparison showed that the critical heat fluxes in the rod bundles with external thermocouple simulators were an average of 13 percent lower at 13.8 MPa, an average of 2.5 percent higher at '0.3 MPa, and an average of 7 percent higher at 6.9 MPa than the critical heat fluxes in the rod bundles without external thermocouple simulators. In the regions of pressure and quality where CHF is predicted to occur in the LOFT core during blowdown operation (about 11 MPa and 30 to 40 percent quality), the magnitudes of CHF between the bundles tested were about the same. Thus, cladding surface thermocouples are

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not expected to have a significant effect on transient CHF or time-to-CHF for the nuclear rods in the LOFT core under blowdown conditions.

A new CHF correlation (designated LOFT-2) was developed, based on the 157 data points from the bundles with rod external thermocouple simulators, to allow a comparison of the data between the bundles with and without rod external thermocouple simulators. However, since a complete data base could not be obtained over the intended range of mass velocities and qualities, the correlation is not adequate to predict CHF for LOFT under blowdown conditions.

The data were compared to predictions based on steady state CHF correlations such as B&W-2, GE and Barnett which are currently used in codes such as RELAP and FRAP-T to predict CHF for LOFT blowdowns. The B&W-2 correlation poorly predicts the magnitude and trend of the data. Most of the data fall outside the applicable range of the B&W-2 correlation. The Barnett correlation severely underpredicts all of the data at mass velocities less than 1360 kg/s-m² and qualities greater than +0.05. The GE correlation provided a good prediction of the data except at mass velocities less than 1360 kg/s-m² and qualities higher than +0.2. In these ranges the GE correlation tends to underpredict almost all of the data.

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I. INTRODUCTION

The Loss-of-Fluid Test (LOFT) facility is an integral nuclear reactor test facility which has been designed to simulate, as nearly as possible all of the important effects that are anticipated to occur during a lossof-coolant accident (LOCA) in a large pressurized water reactor (LPWR) type nuclear steam supply system. LOFT utilizes special experimental instrumentation which has been located to measure the significant interacting nuclear and thermal-hydraulic processes required to assess LOCA and emergency core coolant system (ECCS) performance. The in-core experimental instrumentation includes thermocouples attached to the outside surfaces of the fuel rod cladding. (See Reference 1, Section 2.1.1.2 for details of the thermocouples and attachment.)

An extensive critical heat flux test program has been conducted on electrically heated rod bundles simulating the central region of the LOFT nuclear reactor core. The evaluation and results of initial test data over the pressure range of 13.8 to 16.5 MPa have been previously reported in Reference 2. The results of that data indicate that cladding surface thermocouples reduce the CHF an average of 18 percent over that pressure range.

The purpose of this test program was to determine the effect on CHF due to thermocouples mounted on the cladding surface of some of the LOFT fuel rods over the pressure range of 6.9 to 13.8 MPa in order to:

 determine if rod external thermocouples significantly affect the critical heat flux in the LOFT core for steady state operating conditions in the pressure range where CHF occurs

during blowdown operation and

(2) obtain a data base to develop a CHF correlation to predict critical heat fluxes in the LOFT core for blowdown operation, and to provide data to compare to current steady state CHF correlations used in codes such as RELAP and FRAP-T to predict transient CHF in LOFT under blowdown conditions.

The experimental parameters investigated were:

- (1) Pressures from 6.9 to 13.8 MPa
- (2) Inlet mass velocities from 680 to 3400 kg/s-m²
- (3) Local quality at CHF from -0.25 to +0.60.

Four 25-rod bundles simulating a portion of the Core-1 central (Type A) fuel assembly^[1] in the LOFT reactor were used in the test program. One-hundred-fifty-seven data points were obtained with the rod bundles with rod external thermocouple simulators. These data were compared with 53 data points obtained with the rod bundles without rod external thermo-couple simulators.

This report contains: a description of the CHF test program, test bundles, and test procedure (Section II); a description of the analysis of the data (Section III); a comparison of the data with existing CHF correlations; and a comparison of the data between the rod bundles with and without simulated rod external thermocouples (Section IV); conclusions regarding the effect of thermocouples on CHF conditions in LOFT and adequacy of current models (Section V), and additional tests necessary to obtain the data for developing a CHF correlation applicable for LOFT conditions (Section VI).

II. EXPERIMENT DESCRIPTION

1. SCOPE OF TESTS

CHF data were obtained at the following test conditions:

- (1) Pressure: 6.9, 10.3 and 13.8 MPa
- (2) Mass Velocity: 680 to 3400 kg/s-m²
- (3) Local Quality at CHF: -0.25 to +0.60.

. TEST BUNDLE DESCRIPTION

Two electrically heated test section configurations were selected for the test program. These were:

- (1) A 25-rod bundle in a 5x5 square array with a 1.68 m heated length. This bundle simulated the portion of the Loss-of-Fluid Test (LOFT) reactor central (Type A) fuel assembly as shown in Figure 1^[1]. This asymmetric configuration was chosen to include the effect of rod external thermocouples and rod cluster control (RCC) guide tubes on critical heat flux.
- (2) A 25-rod bundle identical to that described above except that it did not incorporate rod external thermocouple simulators.

Two rod bundles with external thermocouple simulators were built and tested, one by Exxon Nuclear Company (designated EXX2 and ANC1) and one by Aerojet Nuclear Company (designated ANC2)^[a]. Also, two rod bundles without external thermocouple simulators were built (designated ANC3 and ANC4) and tested.

[[]a] As of October 1, 1976, contract being performed by EG&G Idaho, Inc.



2.1 <u>Exxon Rod Bundle Description</u> (bundle with external thermocouple simulators)

This 25-rod bundle (designated EXX2 or ANC1) was selected to investigate the combined effect on CHF of rod external thermocouples, guide tubes, and grid spacers. Its characteristics are described in Table I and a diagram is shown in Figure 2.

2.1.1 <u>Heater Rod Description</u>. The heat generated in the LOFT fuel rods was simulated by passing direct electrical current through rods made of Type 347 stainless steel tubing. Three tube wall thicknesses of 0.287, 0.300, and 0.318 mm were chosen to simulate the radial power profile in the section of the LOFT core modeled by the test section. The normalized power factors of each rod are shown in Figure 2. The tubing outer diameter was 10.7 mm, the same as the fuel rods being simulated. Each rod was filled with ceramic cylinders to prevent collapse under external pressure. The 80-mm long cylinders were machined from 98% dense Al₂O₃ ceramic to fit inside the tubes. The inside of the heated tube was at atmospheric pressure during testing.

Each rod was composed of top and bottom electrodes and a heated section as shown in Figure 3. The heated section was 1.68-m long, the same as the LOFT fuel rods. The top electrode was made of solid nickel with the same OD as the heated section. The top end of the nickel end piece was ground with a self-locking taper which fitted a matching conical hole in the top grid plate. The bottom electrode was joined to copper tubing external to the pressure seal. A 6.4-mm hole through the bottom electrode and ceramic cylinder allowed access for thermocouple instrumentation.

TABLE I

ROD BUNDLE CHARACTERISTICS

Number of rods	25
Number of unheated rods	3
Array	5 x 5
Heated length	1.68 m
Pitch	14.3 mm
Heated Rod OD	10.7 mm
Unheated Rod OD	13.9 mm
Axial heat flux distribution	uniform
Radial heat flux distribution	nonuniform
Number of grid spacers (LOFT-type, no mixing vanes)	5
Grid spacing	0.42 m
Number of simulated external thermocouples on rods with thermocouple simulators	1
Simulated thermocouple OD	1.17 mm
Square flow channel dimension	73.8 mm

Relative power factors.







Figure 3 Heater rod assembly.

2.1.2 <u>Simulated External Thermocouples</u>. Seven of the rods in the test burdle incorporated external thermocouple simulators. Each rod had four simulators. The simulated thermocouples were fabricated from 1.17-mm OD Inconel tubing having a wall thickness of 0.05 mm. This tubing was filled with ceramic spaghetti to prevent collapse under pressure. The simulated thermocouples were laser welded axially along the rod in a manner similar to that of the actual thermocouples on the LOFT fuel rods^[1]. The thermocouple simulators extended nearly the full length on each rod with a simulated thermocouple junction at the same axial level as the actual thermocouple junctions on the LOFT fuel rods. A typical thermocouple attachment and simulated junction are shown in Figure 4.

2.1.3 <u>Simulated Guide Tubes</u>. The test bundle incorporated three unheated rods which simulated RCC guide tubes in the LOFT reactor. These tubes were made of a stainless steel rod with 98% dense $A1_20_3$ ceramic sleeves, 80-mm long with an OD of 13.9 mm, the same as the LOFT RCC guide tubes^[1].

2.1.4 <u>Grid Spacers</u>. The test bundle incorporated five grid spacers (see Figure 5) similar to the LOFT fuel grid spacers^[1]. These spacers did not have mixing vanes, the same as LOFT grid spacers. The spacers had an axial spacing of 0.42 m, the same as in the LOFT core. However, each grid spacer in the test section was located approximately 180 mm upstream of the corresponding LOFT spacer location. This placed the upstream side of the last grid spacer about 150 mm from the end of the heated length (see Figure 6). With this location, CHF occurred both



Figure 4 Photograph of a rod with thermocouple attached.







upstream and downstream of the last spacer, thereby resulting in a mixture of CHF data strongly influenced by a grid spacer (CHF data from downstream of the last spacer) and weakly influenced by a grid spacer (immediately upstream of the last spacer).

Another reason for moving the spacers upstream 180 mm was to facilitate the measurement of the subchannel coolant temperatures at the test section outlet [see subsection 2.1.5 (3)]. The rake of thermocouples could not be installed at the end of the heated length without changing the location of the last grid spacer.

2.1.5 Test Section Instrumentation.

(1) <u>CHF Detection Thermocouples</u>. The heated rods contained internally mounted iron-constantan thermocouples employed as CHF detectors. These thermocouples were located 13 mm upstream from the end of the heated length and 13 mm upstream of the last grid spacer. These thermocouples were brazed to copper washers to improve the detection of temperature changes on the rod surface. This thermocouple installation is shown in Figure 7. The thermocouple leads were brought out of the bottom rod electrode.

(2) <u>Pressure Transducers</u>. The pressure drops across the heated length of the test section and across the grid spacers were measured. The pressure drops were visually read with mercury manometers and also measured with Barton differential pressure cells which were recorded by the data acquisition system. Pressure tap locations are shown in Figure 6.

(3) <u>Cobchannel Coolant Temperatures</u>. Test section outlet subchannel coolant temperatures were measured with a rake of iron-constantan



Dimensions in mm

Figure 7 Single thermocouple instrumentation

thermocouples located 6.4-mm axially downstream from the end of the heated length. These thermocouples were centered within the subchannels.

2.1.6 <u>Flow Housing</u>. The flow housing consisted of four major components: the grid plate, the top adapter, the shroud box, and the bottom adapter.

The grid plate was machined from a nickel plate. It served as a top electrical connection, transferring the current from the rods to a large nickel disc, which in turn led the current out through the insulated pressure seals and to the copper bus. The grid plate was accurately machined to maintain the tops of the rods in proper geometry.

The top adapter located the shroud box with reference to the heated rod geometry and provided a transition from the geometry of the rod bundle channel to an open arrangement for the coolant discharge.

The shroud box was made of four rectangular pieces of stainless steel, machined and fitted to hold the ceramic liner which actually forms the flow channel. The ceramic shroud liner was made of 98% dense Al_2O_3 in 0.5 m long sections, and ground to the desired dimensions within a tolerance of ± 0.08 mm. Holes for pressure tap locations were drilled at selected points along the axial length of the flow channel. The inner dimensions of the installed shroud liner were 73.8 mm square and were sized to allow the enthalpy rise in the peripheral subchannels of the test section to simulate that in adjacent subchannels in the LOFT core.

The bottom adapter located the inlet end of the flow channel with respect to the heated rods, and provided a region for coolant entry into the channel.

2.1.7 <u>"O" Ring Seals</u>. Each rod passed from inside the loop pressure vessel out to the atmosphere through a series of "O" ring seals. This arrangement permitted independent axial movement of each rod.

2.1.8 <u>Electrical Connections</u>. The top electrical connection was made through the grid plate as described earlier. The bottom electrical connections were made by clamping a copper connector to each rod. The connectors were fitted with flexible leads which were connected to an octagonal bus arrangement surrounding the copper electrodes.

2.2 ANC Rod Bundle Description (bundle with external thermocouple simulators)

This 25-rod bundle (designated ANC2) was designed to duplicate the Exxon bundle with external thermocouple simulators. The characteristics of this bundle are the same as those listed in Table I; however, come minor differences existed in the fabrication and instrumentation of this bundle.

The heater rods for this bundle were made of Type 347 stainless steel tubing with wall thicknesses of 0.43, 0.46, and 0.48 mm. The wall thicknesses of these rods were increased to add integrity to the bundle and also to reduce the voltage drop across the grid spacers, thereby reducing the electrical arcing between the grid spacer dimples and the rods and increasing the life of the bundle. Also, the grid spacer dimples and spring tabs were plasma sprayed with a thin coating of Al_2O_3 to reduce arcing between the grid spacers and the rods.

The center rod in this bundle was changed to a high powered rod as compared to a medium powered rod in the Exxon bundle. This was done in an attempt to force CHF to occur in the center of the bundle away from the unheated walls of the test section. The normalized rod power factors for this bundle are shown in Figure 8.

The CHF detecting thermocouples in this bundle were located 13-mm upstream from the end of the heated length and 6.4-mm upstream of the last grid spacer. Four of the rods had several Chromel-Alumel thermocouples located azimuthally around the rod at the axial level just upstream of the last grid spacer as shown in Figure 9. The purpose of this additional instrumentation was to attempt to pinpoint the location of CHF initiation to he'p determine whether or not the external thermocouples have an effect on CHF. The installation of these thermocouples is shown in Figure 10.

2.3 <u>ANC Rod Bundle Description</u> (bundles without external thermocouple simulators)

These 25-rod bundles (designated ANC3 and ANC4) were designed to duplicate the ANC2 rod bundle except that they did not incorporate external thermocouple simulators. The characteristics of these bundles were the same as those listed in Table I except for the thermocouple simulators. The fabrication of the ANC3 bundle was the same as for the ANC2 bundle. The normalized rod power factors for this bundle are shown in Figure 11.

The ANC3 bundle was instrumented with azimuthal thermocouples just upstream of the last grid spacer as shown in Figure 12.



Figure 8 Configuration of rod bundle with external thermocouple simulators (ANC2).



Figure 9 Azimuthal location of CHF detecting thermocouples in ANC2 test bundle.



Figure 10 Installation of azimuthal CHF detecting thermocouples.

Relative power factors The power factors in parentheses are for runs 93 to 108. Four rods, were replaced for these num







Notes: Axial Level 6.35 mm Upstream of Last Grid Spacer (0.152 m upstream from end of heated length)

View of Bundle Outlet

Figure 12 Azimuthal location in CHF detecting thermocouples in ANC3 test bundle.

The ANC3 bundle provided data at 13.8 MPa but failed prior to testing at 6.9 and 10.3 MPa.

The 25-rod bundle (designated ANC4) was used to obtain data at 6.9 and 10.3 MPa. The ANC4 bundle configuration was the same as the ANC3 bundle. The normalized rod power factors for the ANC4 bundle are shown in Figure 13. This bundle was primarily designed for transient (blowdown) CHF tests so the internal CHF sensing thermocouples were located differently than for the ANC3 bundle. Only one thermocouple was positioned in each rod at 12.7 and 152.4 mm from the end of the heated length. The azimuthal locations of these thermocouples are shown in Figure 14. These thermocouples were mounted in the ceramic plugs in the rods, the same as the azimuthal thermocouples in the ANC2 and ANC3 bundles.

Relative power factors




✗ Internal T/C Locations for Measuring CHF

Figure 14 Azimuthal location of CHF detecting thermocouples in ANC4 test bundle

3. EXPERIMENT PERFORMANCE

Forty of the 157 data points for the rod bundles with rod external thermocouple simulators were obtained with the EXX2 bundle in 1973. These data points were taken at 10.3 and 13.8 MPa. The remaining 117 data points were obtained with the ANC2 bundle in November 1975. These data were taken at 6.9, 10.3 and 13.8 MPa.

Nineteen of the 53 data points for the rod bundles without rod external thermocouple simulators were obtained with the ANC3 bundle in December 1975. The ANC3 bundle failed beyond repair at this point (see Reference 2). These data were taken at 13.8 MPa. The remaining 34 data points were obtained with the ANC4 bundle in May 1977. Most of these data were taken at 6.9 and 10.3 MPa with only 3 data points taken at 13.8 MPa.

Pressure drop tests were conducted on all 4 of these bundles. The data were in excellent agreement.

III. ANALYSIS OF DATA

1. DISCUSSION

The data reduction method to be explained was used for all four sets of CHF data: EXX2, ANC2, ANC3 and ANC4. Any variations in the method for the different sets are explained fully. The objective of the data reduction was to arrive at accurate subchannel coolant conditions at each CHF location for each test run. This allowed a CHF correlation using local subchannel coolant conditions to be developed or the data to be compared with existing steady state CHF correlations.

The test data from the four sets of CHF tests were in the form of computer printouts of conditions during each run, strip charts indicating rod CHF occurrence, and electrical resistance measurements for the heated rods. The run conditions included the test section coolant inlet temperature, system pressure, flow rate, power level and pressure drop across various segments of the test bundle. The power was determined from voltage (bus-to-bus) and current measurements for the bundle. Each heated rod in a bundle had a 1.08 m segment of stainless statl, termed the heated length, and nickel and copper extensions at both ends. The bus-to-bus voltage measurement included the voltage drop across the nickel and copper extensions. In order to determine the power for the heated length portion of the test bundle, a voltage correction factor was used

 $F = \frac{V_{bus-to-bus}}{V_{heated length}}$

where V = voltage.

Assuming the voltage across the heated length was the same for all of the heated rods, the power was corrected by

Pheated length = Pbus-to-bus/F

where P = power.

The heated length voltage was measured during the ANC3 tests, and the results, F as a function of inlet temperature, are shown in Figure 15. The ANC2, ANC3, and ANC4 power measurements were corrected using this function. The curve of F versus inlet temperature (Figure 15) was not applicable to the EXX2 bundle because of the differences in the heater rod wall thickness between the ANC2, ANC3, and ANC4 bundles, and the EXX2 bundle. For the EXX2 data the voltage correction factor was assumed to be a constant 1.025, regardless of inlet temperature because no $V_{heated length}$ measurements were available for these tests.

The inlet enthalpy, needed as input to COBRA-II, was determined from the inlet temperature and system pressure using the ASME Steam Tables^[3]. The resulting run condition data for the four CHF tests are listed in Appendix A.

Internal thermocouple voltage readings during each test were recorded on strip charts. The occurrence of CHF near an internal thermocouple location resulted in a sharp rise in the thermocouple voltage reading. A typical strip chart indication of CHF is shown in Figure 16. From the charts, the rod(s) experiencing CHF was determined for each test run. When there were multiple CHF indications, either at different axial levels on one rod or on different rods, all indications were assumed to be independent CHF indications.



Figure 15 Voltage correction, F, versus inlet temperature for the ANC2, ANC3 and ANC4 test sections.

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Figure 16 Thermocouple strip chart traces showing typical CHF indications.

CHF indications occurring significantly after the initial CHF indication were not included in the data base because they were usually the result of a system pressure drop when the power was reduced.

Once the CHF rod(s) was identified for each test run, a corresponding subchannel was chosen. Most of the ANC2 and ANC3 CHF indications came at axial locations in which the azimuthal thermocouples indicated the azimuthal CHF region^[a]. If CHF occurred at the end of the bundle (1.68 m) where there were no azimuthal thermocouples, CHF was assumed to occur at the same azimuthal region as indicated by the thermocouples at the 1.52 m level. In cases where there was some doubt as to CHF region, one of the cooler subchannels adjacent to the CHF rod was chosen in order to be conservative. The EXX2 test was run with nondirectional thermocouples. CHF regions were chosen so that they agreed with those chosen for similar CHF indications in the ANC2 bundle.

With the system conditions for each CHF run and the location of CHF known, the subchannel code COBRA-II was used to predict the local subchannel conditions at CHF, such as mass flux and equilibrium quality.

[[]a] At times, two directional thermocouples would indicate CHF on the same rod at the same axial level, but in adjacent subchannels. In such cases, both indications were considered as independent CHF points. Actually, the CHF was probably occurring between the two subchannels.

2. SUBCHANNEL CODE AND MODEL

2.1 COBRA-II Description

The steady state subchannel analysis code COBRA-II^[4] was used in the data analysis. COBRA-II "..., computes the flow and enthalpy in the subchannels of rod bundle nuclear fuel during both boiling and nonboiling conditions. It uses a mathematical model that considers two types of crossflow mixing. The first type is a net diversion crossflow caused by flow redistribution, and the second is a turbulent (fluctuating) crossflow caused by the random travel of coolant between adjacent subchannels. Each subchannel is assumed to contain one-dimensional, twophase slip-flow and is assumed to have a sufficiently fine two-phase flow structure to permit specification of local void fraction as a function of enthalpy, flow, pressure and position. The equations of the mathematical model are solved as an initial-value problem by using finite differences."^[4] COBRA-II was used rather than the newer version COBRA-IIIC^[5], which also handles time-dependent problems, because of the difference in execution times. Both programs give essentially the same results for steady state problems. The program is listed in Reference 2.

2.2 Model of Test Bundle

Figure 8 shows the COBRA-II model of the test sections. There were 22 heated rods of 10.7 mm diameter, three unheated rods of 13.9-mm diameter, and an unheated shroud. The five grid spacer locations are shown in Figure 6. The COBRA-II calculations were started at the beginning of the heated lengths of the rods. Uniform enthalpy and mass flux were assumed at this level. This is a good assumption because there is a

significant length of straight flow and one grid spacer upstream of the beginning of the heated length.

2.3 Single-Phase Friction factor

The flow through the test bundle for the range of test conditions covered in the CHF test was turbulent. Therefore, a friction factor of the form

$$f = CRe^{-0.20}$$

where Re = Reynolds number for the test section, was used^[6]. ΔP predictions for the highly subcooled test runs of ANC2 resulted in the value 0.213 for C. It should be noted that the ΔP comparisons were made only across the total heated length. Shorter incremental ΔP measurements were judged to be highly uncertain. The value 0.213 is slightly higher than the 0.203 used in previous data evaluations^[7,8]. It is believed that 0.213 is more accurate than 0.203 because the data base used in this analysis is much larger than that used previously. The value 0.213, 16% above the smooth tube figure of 0.184, is not unusual. Tong and Weisman^[9] indicate that for a bundle with a pi⁺ch-to-diameter ratio of 1.27 (the CHF bundles had a ratio of 1.33), the friction factor is 0 to 20% higher than the smooth tube correlation.

2.4 Grid Spacer Coefficients

COBRA-II computes the $\triangle P$ across grid spacers by using user-defined coefficients for each spacer and subchannel within the spacer. The coefficients are defined as:

$$C = \frac{\Delta P}{\frac{1}{2} \frac{\rho}{g_c} \sqrt{2}}$$

where

v = velocity of the fluid

 ρ = density of the fluid

g_= Newton constant.

The spacer coefficients used for this work came from the results of velocity and pressure drop measurements made on a 15 x 15-rod LOFT central fuel assembly^[7]. The spacers for the 15 x 15-rod test were similar to those for the CHF tests. For each spacer, four different spacer coefficients were used, depending on the type of subchannel:

$C = 2.91 \text{ Re}^{-0.142}$	Corner subchannels
$C = 2.25 \text{ Re}^{-0.142}$	Outer subchannels (bordering on
	shroud inner wall)
$C = 4.20 \text{ Re}^{-0.142}$	Inner subchannels
$C = 4.99 \ Pe^{-0.142}$	Inner subchannels adjacent to a
	simulated guide tube

where Re = Reynolds number for the subchannel.

In the 15 x 15-rod test large variations in flow were present at the inlet, so the lowest two sets of spacer coefficients were different from the others because of large flow redistributions. In the CHF tests the inlet flow was essentially uniform, so all of the spacers were assumed to have the same set of coefficients as Spacers 3 through 5 in Reference 7.

2.5 Turbulent Mixing Coefficient

COBRA-II models turbulent mixing momentum and enthalpy interchange between adjacent subchannels by the use of a turbulent mixing coefficient $\beta^{[10]}$. Hot mixing results obtained before the CHF tests included outlet coolant temperature measurements for each subchannel. An evaluation of B using the hot mixing data resulted in

$$\beta = 0.0062 \frac{D}{S} \text{Re}^{-0.10}$$

where

D = average hydraulic diameter for the two subchannels in question S = rod gap spacing between the two subchannels in question Re = average Reynolds number for the two subchannels in question.

However, there was very little difference in the COBRA-II predictions over the range $B = (0.00 \text{ to } 0.0120) \frac{D}{S} \text{ Re}^{-0.10}$. These results agree with previous estimates of $B^{[7,11]}$ for LOFT Core-1 fuel assemblies.

2.6 True Quality Correlation

COBRA-II has options for various two-phase correlations for true quality, void fraction, and friction multiplier. For true quality, the Levy model was used (in the program it is termed a subcooled void correlation, but it actually predicts only the true quality). Weisman and Bowring^[12] believe the Levy model is more accurate than modified versions of the Bowring model, used in the subchannel codes HAMBO^[13] and some versions of THINC^[14], for mass fluxes greater than 680 kg/s-m² [the range for the CHF tests was 680 - 3400 kg/s-m²]. Also Collier^[15] states that

the Levy correlation agrees well with the data of Egan, of Maurer, and of Rouhani.

2.7 Void Correlation

The Thom subcooled void correlation modified by Tong^[9] was used for the subcooled region, and the Thom annular void model^[9] was used for the positive equilibrium quality region. Several other void correlations investigated were the modified Armand^[16], the homogeneous^[4], the Levy, and the Bowring^[15]. The Thom correlations were chosen because they generally predicted lower void fractions than the other models, which reduced the predicted pressure drops and improved the agreement with the ANC2 ΔP data. Castellana and Bonilla^[17] indicate that the Thom and Bowring correlations are widely used and are generally recognized to show good agreement with available data. Also, Tong and Weisman^[9] feel that the Thom subcooled correlation is somewhat more accurate than the Bowring model. These observations and the fact that the Thom correlations resulted in the most accurate ΔP predictions are justification for using the Thom models for the CHF data reduction.

2.8 Two-Phase Friction Multiplier

The Armand two-phase friction multiplier^[4] was used in COBRA-II. Several other correlations investigated were the Thom, Levy, Baroczy, homogeneous (in COBRA-II), and Martinelli-Nelson^[15]. The Armand Correlation generally predicted the lowest friction multipliers resulting in lower ΔP predictions, which agreed best with the , NC2 ΔP data.

For bulk boiling, the region in which the friction multiplier is the largest and has the most effect on the ΔP predictions, the Armand, Thom, and homogeneneous models are similar^[17] and any of these three could have enused. The Martinelli-Nelson model was not used because it tends to overpredict the value of the friction multiplier for most of the mass fluxes involved, 680 to 3400 kg/s-m².

2.9 Pressure Drop Predictions

Pressure drop data were obtained for each run during the CHF tests. Although six ΔP values were recorded, corresponding with various lengths of the test section, only the ΔP across the total heated length of the bundle was compared with the COBRA-II predictions. The uncertainty in the other five measurements was large.

A comparison of measured to predicted test section pressure drops was made. The predicted pressure drops were generally lower than measured in the ,uality range of 0 to 30% and generally higher than measured in the quality range above 30%. The pressure drop predictions were an average of 4.6% low for all of the ANC4 runs and 2.3% low for all of the ANC2 runs. The change in the trend of measured to predicted pressure drop near a quality of 30% could be due to a change in flow regime from bubbly-slug flow to annular flow.

2.10 Power Generation In Rods

2.10.1 <u>Rod Power Factors</u>. The rod power factor determinations are an important part of the data reduction because they directly affect

the experimental critical heat flux values and the COBRA-II predictions of local CHF conditions. Rod power factors represent the power generation in a rod relative to the average rod power generation in a test bundle. They were calculated from rod resistances measured at Columbia University before and after CHF testing. The conversion of resistances to power factors involved several steps and involved mostly simple electrical paraliel and series resistance formulas. The first step was to determine the resistances at the test temperatures, which were assumed to be 640 K for the heated stainless steel sheaths and 590 K for the nickel extensions. The copper part of the extensions contributed a negligible amount to the resistance of the extensions. Then, power factors for the rods (heated length and extensions) were calculated using the formula

$${}^{\text{pF}}\mathsf{T},i = \frac{\mathsf{R}_{eq,T}}{\mathsf{R}_{T,i}}$$

where

$$R_{eq,T} = \frac{1}{\sum_{j=1}^{22} \frac{1}{R_{T,i}}}$$

The subscript T denotes total rod (heated length and extensions) and i denotes the ith rod. All resistances used were those determined for the run temperatures mentioned previously.

Finally, the power factors were corrected to reflect only the power generated in the heated length of the rods. The formulas used are

$$PF_{HL,i} = PF_{T,i} \left(\frac{R_{HL,i}}{R_{T,i}} \right)$$
 (constant)

22 =
$$\left(\sum_{j=1}^{22} PF_{HL,j}\right)$$
 (constant)

The subscript HL denotes heated length. All resistances used were those appropriate for the run temperatures. The original rod resistances at room temperature, intermediate calculations, and the final rod power factors are listed in Tables II through V for the four sets of CHF tests.

2.10.2 External Thermocouple Effect on Power Factors. Some of the power generated in a rod with external thermocouple simulators was actually generated in the simulators. This distinction was not needed for the COBRA-II calculations because predictions of enthalpy at the region of CHF do not distinguish between rod and thermocouple power generation. However, when determining the rod heat flux at the CHF location, the thermocouple power generation should not be included. The heat flux at the rod surface was much greater than at the thermocouple simulator surface. Therefore, the CHF probably started on the rod surface rather than on the simulator. In order to determine the surface heat flux at the point of CHF, the power generated by the thermocouple simulators was eliminated, and then the remaining power was assumed to be spread uniformly on the rod surface (excluding the thermocouple simulator surfaces).

It should be noted that some of the total power generation in a rod with external thermocouples also occurred within the welds attaching the thermocouples to the rod. Exxon Nuclear Company estimated that about

Rod Number	Resistances ~ 300 K (78°F) Heated Length (Ω) Extensions (Ω)		Resistances - Run Heated Length (2)	Temperatures[a] Extensions (Ω)	Rod Power Factors	Rod Power Factors With Thermocouple Effect Eliminated	
1	0.1313	0.00090	0.1804	0.00287	0.945		
2	0.1334	0.00100	0.1792	0.00319	0.948		
3	0.1256	0.00090	0.1667	0.00287	1.020	0.967	
4	0.1328	0.00100	0.1784	0.00319	0.953		
5	0.1255	0.00075	0.1665	0.00303	1.020	0.967	
6	0.1335	0.00095	0.1795	0.00303	0.949		
7	0.1247	6.00100	0.1675	0.00319	1.012		
8	0.1150	0.00100	0.1526	0.00319	1.107	1.053	
9		-		사람이 있는 것 같은 것이 같다.			
10	0.1249	0.00120	0.1677	0.00382	1.004		
11	0.1266	0.00200	0.1700	0.00637	0.962		
12	0.1248	0.00120	0.1676	0.00382	1,005		
13			- 1570	0.00543	0.000		
14	0.1245	0.00170	0.1672	0.00341	0.966		
15	0.1252	0.00100	0.1681	0.00319	0.009		
16	0.1325	0.00110	0.1779	0.00350	0.952	0 963	
17	0.1258	0.00100	0.1669	0.00319	1.010	130.0	
18	0.1274	0.00100	0.1691	0.00319	1.005	230 0	
19	0.1263	0.00090	0.10/0	0.00207	0.002	0.502	
20	0.1276	0.00090	0.1714	0.00207	1 010		
21	0,1239	0.00100	0.1004	0.00319	1.012		
22		0.00005	0 1600	0 00271	2 010		
23	0.1258	0.00065	0.1710	0.00287	0 996		
24	0.1273	0.00090	0.1563	0.00287	1.074	1.021	
25	0.1178	0.00120	0.1303	0.00302	1.914		
1-1-	the latter was as an an end of the second		and a second		ADDED TO ADD	and the second	

TABLE II

ROD RESISTANCES AND RELATIVE POWER FACTORS FOR THE EXX2 TEST BUNDLE

[a] Run temperatures are defined as 640 K (700°F) for the heated lengths and 590 K (600°F) for the extensions.

39

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Rod Number	Resistances - 300 K (78°F) Heated Length (0) Extensions (0)		Resistances - Run Heated Length (R)	Temperatures[a] Extensions (D)	Rod Power Factors	Rod Power Factor With Thermocoupl Effect Eliminate	
1 2 3 4 5	0.08660 0.08605 0.08225 0.08675 0.08245	0.00185 0.00170 0.00175 0.00170 0.00180	0.11630 0.11557 0.10964 0.11651 0.10991	0.00589 0.00541 0.00557 0.00541 0.00543	0.933 0.946 0.990 0.939 0.985	0.954	
6 7 8 9	0.08720 0.07810 0.07350	0.00160 0.00140 0.00195	0.11711 0.10489 0.09805	0.00510 0.00446 0.00621	0.939 1.051 1.081	1.045	
10 11 12	0.07970 6.08200 0.07780	0.00175 0.00175 0.00185	0.10704 0.11013 0.10449	0.00557 0.00557 0.00589	1.011 0.986 1.027		
13 14 15 16 17 18 19 20 21	0.07785 0.07975 0.08865 0.08215 0.08195 0.08205 0.08100 0.07760	0.00170 0.00160 0.00185 0.00170 0.00155 0.00190 0.00190 0.00190	0.10455 0.10710 0.11637 0.10951 0.10924 0.10937 0.10878 0.10422	0.00541 0.00510 0.00589 0.00541 0.00494 0.00605 0.00605 0.00605	1.036 1.019 0.933 0.993 1.004 0.984 0.988 1.045	0.957 0.5 3 0.949	
22 23 24 25	0.07800 0.08160 0.07385	0.00155 0.00170 0.00195	0.10475 0.10959 0.09852	0.00494 0.00541 0.00621	1.043 0.993 1.076	1.040	

TABLE III

ROD RESISTANCES AND RELATIVE POWER FACTORS FOR THE ANC2 TEST BUNDLE

[a] Run temperatures are defined as 640 K (700°F) for the heated lengths and 590 K (600°F) for the extensions

40

Rod Number	Resistance Heated Length (es - 300 K (78°F) Ω) Extensions (Ω)	Resistances Heated Length	- Run Temperatures[a] (Ω) Extensions (Ω)	Rod Power Factors
1 2 3 4 5 6 7 8	0.08630 0.08655 0.08650 0.08725 0.08625 0.08710 0.07810 0.07810 [b]	0.00175 0.00155 0.00170 0.00155 0.00165 0.00170 0.00165 0.00170(0.00175)	0.11590 0.11624 0.11617 0.11718 0.11583 0.11698 0.10489 0.10402(0.10704)	0.00557 0.00494 0.00541 0.00494 0.00526 0.00541 0.00526 0.00541(0.00557)	0.950 (0.952) 0.957 (0.959) 0.951 (0.952) 0.950 (0.952) 0.956 (0.957) 0.945 (0.946) 0.046 (0.047) 1.051 (1.023)
9 10 11 12	0.07795 0.08170 0.07780(0.07785)	0.00160 0.00140 0.00155(0.00170)	0.10469 0.10972 0.10449(0.10455)	0.00510 0.00446 0.00494(0.00541)	1.051 (1.052) 1.018 (1.020) 1.055 (1.047)
13 14 15 16 17 18 9 20 21	0.07735 0.07830 0.08640 0.08650 0.08670 0.08655 0.08160 0.08160 0.07795(0.07760)	0.00180 0.00145 0.00160 0.00170 0.00175 0.00175 0.00170 0.00180 0.00160(0.00160)	0.10328 0.10516 0.11604 0.11617 0.11644 0.11624 0.10959 0.10469(0.10422)	0.00573 0.00462 0.00510 0.00541 0.00557 0.00541 0.00573 0.00573 0.00510(0.00510)	1.046 (1.047) 1.055 (1.057) 0.956 (0.958) 0.951 (0.952) 0.946 (0.948) 0.950 (0.952) 0.997 (0.998) 1.051 (1.057)
22 23 24 25	0.07795(0.07800) 0.08205 0.07805	0.00155(0.00155) 0.00130 0.00165	0.10469(0.10475) 0.11019 0.10487	0.00494(0.00494) 0.00414 0.00526	1.054 (1.055) 1.020 (1.021) 1.046 (1.048)

TABLE IV

ROD RESISTA" .ES AND RELATIVE POWER FACTORS FOR THE ANC3 TEST BUNDLE

[a] Run temperatures are defined as 640 K (700°F) for the heated lengths and 590 K (600°F) for the extensions.

[b] All quantities in parentheses indicate changes for CHF runs 93 to 108 caused by the replacement of four rods.

Rod	Resistances - 3	00 K (78°F)	Resistances - Run	Rod Power	
Number	Heated Length (Ω)	Extensions (1)	Heated Length (Ω)	Extensions (2)	Factors
1	0.08645	0.00155	0.11610	0.00494	0.949
2	0.08605	0.00150	0.11557	0.00478	0.956
3	0.08620	0.00145	0.11577	0.00462	0.957
4	0.08650	0.00145	0.11617	0.00462	0.954
5	0.08640	0.00145	0.11604	0.00462	0.955
6	0.08630	0.00155	0.11590	0.00494	0.950
7	0.07735	0.00145	0.10268	0.00462	1.057
8	0.07905	0.00150	0.10616	0.00478	1.033
9		-	-	-	-
10	0.07730	0.00150	0.10381	0.00510	1.049
11	0.08080	0.00160	0.10851	0.00510	1.007
12	0.07755	0.00150	0.10415	0.00478	1.052
13		-		-	-
14	0.07765	0.00150	0.10428	0.00478	1.050
15	0.07715	0.00150	0.10361	0.00478	1.056
16	0.08610	0.00155	0.11563	0.00494	0.953
17	0.08650	0.00145	0.11617	0.00462	0.954
18	0.08565	0.00160	0.11503	0.00510	0.955
19	0.08595	0.00150	0.11543	0.00478	0.957
20	0.08165	0.00150	0.10966	0.00478	1.003
21	0.07760	0.00140	0.10422	0.00446	1.057
22		-		-	*
23	0.07640	0.00195	C.10261	0.00621	1.038
24	0.08160	0.00145	0.10959	0.00462	1.007
25	0.07705	0.00155	0.10348	0.00494	1.055

TABLE V										
00	RESISTANCES	AND	RELATIVE	POWER	FACTORS	FOR	THE	ANC4	TEST	BUNDLE

[a] Run temperatures are defined as 640 K (700°F) for the heated lengths and 590 K ($600^{\circ}F$) for the extensions.

2% of a rod power generation occurred in the welds for the EXX2 bundle. This resulted in the estimate that the heat flux at the weld surfaces was twice that of the rod surface. A similar calculation for the ANC2 bundle indicated that the heat flux at the weld surfaces was less than at the rod surface. The reason for the difference between bundles is that the welds on the ANC2 bundle had much larger surface areas than did those on EXX2 and ANC1. Analysis of the CHF results (see Reference 2) indicated that no significant difference between bundles could be attributed to the difference in heat flux at the welds.

Assuming parallel electrical resistances for a rod and its four external thermocouple simulators, a formula for the total thermocouple simulator power generation in a rod was derived^[a]

$$4 P_{T/C} = P_{W} \left(1 - \frac{R_{T/C}}{R_{T/C} + 4R_{W/O}}\right)$$

where P = power. The subscript T/C denotes one thermocouple simulator, w means a rod with thermocouple simulators, and w/o, an identical rod without thermocouple simulators. All quantities refer to the reated lengths only.

Using the preceding formula and assuming a test temperature of 640 K for the rod and thermocouple simulators resulted in a 3.3% power generation for the thermocouple simulators on Rods 8 and 25, and a 3.6% power generation for Rods 3, 5, 17, 18 and 19 for the ANC2 bundle. Since Rods 8 and 25 were high power rods, and their rod resistances were lower than the other rods, the thermocouple simulator effect was correspondingly smaller. Using the same formula for the EXX2 bundle

[[]a] It was assumed that whatever power was generated in the welds attaching the thermocouple simulator to the rod wall could be treated as though it were generated uniformly in the rod.

resulted in a 4.9% power generation in the simulators for Rods 8 and 25, and a 5.2% power generation for Rods 3, 5, 17, 18, and 19. Since the rod walls in the EXX2 bundle were thinner than those for ANC2, the thermocouple simulator power generation values were larger than those for ANC2. The resulting rod power factors with the thermocouple simulator power generation subtracted are listed in Table II for EXX2 and Table III for ANC2. ANC3 and ANC4 bundles had no thermocouple simulators, so the rod power factors for them are unchanged.

xon Nuclear Company^[19] estimated, by another method, the thermocouple simulator power generation to be 4.6% for the EXX2 bundle using room temperature resistances. The equation in this report with room temperature resistances predicts 4.5% for the same rods. The major differences between Reference 19 and this analysis are that in this report an attempt was made to account for resistance changes when the bundles were heated up to actual run conditions and to account for minor differences between high power rods (8 and 25) and low power rods, (3, 5, 17, 18, and 19).

3. SUBCHANNEL RESULTS

The conditions for each test run and used as input to the COBRA II code are listed in Appendix A for the EXX2, ANC1, ANC2, ANC3, and ANC4 data sets.

The results of the data reduction work, critical heat flux, local quality, and mass flux for each CHF occurrence are listed in Appendix B for all four sets of tests.

IV. RESULTS

1. COMPARISON OF DATA BETWEEN BUNDLES WITH ROD EXTERVAL THERMOCOUPLE SIMULATORS (EXX2 AND ANC2) AND BUNDLES WITHOUT RUD EXTERNAL

THERMOCOUPLE SIMULATORS (ANC3 AND ANC4)

The data from the bundles with rod external thermocouple simulators (EXX2, ANC1 and ANC2) have been compared with the data from the bundles without rod external thermocouple simulators (ANC3 and ANC4). In order to make this comparison, a new CHF correlation was developed (designated LOFT-2) based on the 157 data points from the bundles with rod external thermocouple simulators. The data base for the LOFT-2 correlation does not completely cover all combinations of pressure, quality and mass velocity over the pressure range of 6.9 to 13.8 MPa, however, the correlation does provide a valid means of comparing the data between the bundles because the data for the bundles with and without external thermocouple simulators were at approximately the same local fluid conditions. Since the data base for the LOFT-2 correlation is limited, the correlation does not provide a valid prediction of CHF for all combinations of pressure, quality and mass velocity that might occur in the LOFT reactor during blowdown. Therefore, the LOFT-2 correlation is not used for that application. The LOFT-2 correlation, however, satisfactorily predicts the 157 data points from the EXX2, ANC1 and ANC2 data sets as shown in Figures 17 through 20 with a mean value of measured to predicted CHF of 0.988 and a standard deviation of 0.106.

The data from the bundles without rod external thermocouple simulators (ANC3 and ANC4) have been compared with the LOFT-2 correlation as shown in Figures 21 through 24. Figure 21 indicates that data from the bundles without rod external thermocouple simulators is basically the

ANC1, ANC2, EXX2



Figure 17 Comparison of measured to predicted critical heat flux. (LOFT-2 + COBRA-II) for LOFT bundles with external thermocouple simulators.



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Figure 18 Ratio of measured to LOFT-2 predicted critical heat fluxvs-pressure for LOFT bundles with external thermocouple simulators.

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ANC1, ANC2, EXX2



Figure 19 Ratio of measured to LOFT-2 predicted critical heat flux-vsmass velocity for LOFT bundles with external thermocouple simulators.

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simulators.

ANC1, ANC2, EXX2

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Figure 21 Comparison of measured to predicted critical heat flux. (LOFT-2 + COBRA-II) for LOFT bundles without external thermocouple simulators.



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Figure 22 Ratio of measured to LOFT-2 predicted critical heat flux-vspressure for LOFT bundles without external thermocouple simulators.

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ANC3, ANC4



Figure 23 Ratio of measured to LOFT-2 predicted critical heat flux-vsmass velocity for LOFT bundles without external thermocouple simulators.

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ANC3, ANC4

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same as the data from the bundles with rod external thermocouple simulators over the pressure range of 6.9 to 13.8 MPa. The mean value of measured to predicted CHF with the LOFT-2 correlation for these 53 data points is 1.015. Figures 22 and 24 indicate there is a slight trend with pressure and quality such that the data from the bundles with rod external thermocouples are an average of 13 percent lower at 13.8 MPa, an average of 7.5 percent higher at 10.3 MPa and an average of 7 percent higher at 6.9 MPa than the data from the bundles without rod external thermocouple simulators. This is consistent with earlier data^[2] which showed that CHF in the bundles without the thermocouple simulators was increasingly higher than CHF in the bundles with the thermocouple simulators as the pressure increased above 13.8 MPa.

Although the data is limited over the ranges of pressure, mass velocity and quality tested, the range of the data is sufficient to draw conclusions regarding the effect of rod external thermocouple simulators on CHF in the bundles tested. Based on the test results, rod external thermocouple simulators do not have a significant effect on the magnitude of CHF in the bundles tested near 11 MPa and 30 to 40 percent quality, which is where CHF is predicted to occur during a blowdown in LOFT from 15.5 MPa. Thus, cladding surface thermocouples on LOFT fuel rods are not expected to have a significant effect on the transient critical heat flux or time-to-CHF in the LOFT core under blowdown conditions.

2. COMPARISON OF DATA FROM THE BUNDLES WITH ROD EXTERNAL THERMOCOUPLE SIMULATORS (EXX2 AND ANC2) WITH VARIOUS STEADY STATE CHF CORRELATIONS

The data from the bundles with external thermocouple simulators (EXX2 and ANC2) have been compared with the predicted critical heat fluxes of various steady state CHF correlations which are being used in codes such as RELAP and FRAP-T to predict CHF in the LOFT core under blowdown conditions. The ability of these correlations to predict CHF under blowdown conditions in the LOFT reactor has not been verified.

In Figures 25 through 28 the measured critical heat fluxes are compared with the predicted critical heat fluxes of the B&W-2 correlation (see Appendix C). This correlation overpredicts the data at qualities less than +0.10 and underpredicts the data at qualities greater than +0.15 and mass velocities less than 1360 kg/s-m². However, most of the data is outside the applicable range of the B&W-2 correlation.

Figures 29 through 32 show a comparison of the measured critical heat fluxes with the predicted critical heat fluxes of the Barnett correlation (see Appendix C). This correlation underpredicts almost all of the data, particularly at qualities higher than +0.10 and mass velocities less than 1632 kg/s-m².

Figures 33 through 36 show a comparison of the measured critical heat fluxes with the predicted critical heat fluxes of the General Electric CHF correlation (see Appendix C). This correlation does a good job of predicting the LOFT data except at qualities higher than +0.20 and mass velocities less than 1360 kg/s-m². The correlation underpredicts the data at those conditions.



ANC1, ANC2, EXX2

Figure 25 Comparison of measured to predicted critical heat flux. (B&W-2 + COBRA-II) for LOFT bundles with external thermocouple simulators.



ANCI, ANC2, EXX2



Ratio of measured to B&W-2 predicted critical heat flux-vs-mass velocity for LOFT bundles with external thermocouple simulators. Figure 27



ANCI, ANC2, EXX2
ANC1, ANC2, EXX2



Figure 29 Comparison of measured to predicted critical heat flux. (Barnett + COBRA-II) for LOFT bundles with external thermocouple simulators.





ANC1, ANC2, EXX2



Figure 31 Ratio of measured to Barnett predicted critical heat flux-vsmass velocity for LOFT bundles with external thermocouple simulators.

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simulators.

ANC1, ANC2, EXX2



ANC1, ANC2, EXX2

Figure 33 Comparison of measured to predicted critical heat flux. (GE + COBRA-II) for LOFT bundles with external thermocouple simulators.



Ratio of measured to GE predicted critical heat flux-vs-pressure for LOFT bundles with external thermocouple simulators.

ANC1, ANC2, EXX2



Figure 35 Ratio of measured to GE predicted critical heat flux-vs-mass velocity for LOFT bundles with external thermocouple simulators.

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MEASURED CRITICAL HEAT FLUX /

3. LOCATIONS OF CHF AND POSTULATED MECHANISMS

3.1 Bundles with Rod External Thermocouple Simulators

For the LOFT CHF tests at 13.8 to 16.5 MPa reported in Reference 2, 449 of the 455 data points from the bundles with rod external thermocouple simulators (EXX2, ANC1 and ANC2) were obtained on rods with the thermocouple simulators. The majority of this data was in the subcooled region (local quality < 0). The remaining six data points were obtained on rods that did not have thermocouple simulators. All six of these data points were obtained at local qualities higher than +10 percent.

For the EXX2 and ANC2 bundles at 13.8 MPa, 68 of 71 data points were obtained on rods with external thermocouple simulators. Sixty-three of the 71 data points were also at local qualities less than +10 percent. The remaining three data points which occurred on rods without external thermocouple simulators occurred at local qualities higher than +15 percent.

For the EXX2 and ANC2 bundles at 6.9 and 10.3 MPa, only 12 of 86 data points were obtained on rods with external thermocouple simulators. All but two of these 12 data points were at local qualities less than +10 percent. Of the remaining 74 of the 86 data points obtained on rods withou⁺ external thermocouple simulators, almost all were at local qualities of +10 percent or higher.

It appears that thermocouples mounted externally on the cladding surface of LOFT fuel rods affect the steady state CHF mechanism on the rods they are attached to. The mechanism seems to change near +10 percent quality, probably due to a change in flow regime. For very low positive qualities and subcooled conditions the flow regime is

bubbly flow. It is postulated that small bubbles form in the region directly between a rod with external thermocouple simulators and an adjacent rod. The external thermocouples reduce the ability of these bubbles to be swept away, causing CHF to be initiated at that location and resulting in a reduced magnitude of CHF. This is consistent with the high pressure low quality data reported in Reference 2. In these tests, burn marks were apparent on the rods with external thermocouple simulators in the region directly facing an adjacent rod and between the thermocouple simulators. It was also shown that in the low quality region, the critical heat fluxes in the bundles with rod external thermocouples were significantly lower than the CHF in the bundles without rod external thermocouples.

For fluid qualities higher than +10 percent, the flow regime changes to annular flow. In this flow regime a liquid film forms on the rods and thermocouple simulators. Due to the added surface area of the thermocouple simulators, the rods with the thermocouple simulators are cooled more than the rods without the thermocouple simulators. This would account for the fact that all or the CHF data points from the EXX2 and ANC2 bundles at qualities higher than +10 percent were all shifted to rods without the thermocouple simulators, even though these rods had lower power factors than the rods with the thermocouple simulators. Although the data points were shifted to the rods without thermocouple simulators in the EXX2 and ANC2 bundles, the critical heat fluxes were about the same magnitude as the critical heat fluxes in the ANC3 and ANC4 bundles in the high quality region.

The critical heat fluxes in the LOFT core under blowdown conditions will occur at high quality dry out conditions. The results of this test indicate that the LOFT rods with cladding surface thermocouples could have additional cooling due to the thermocouples and potentially shift the critical heat flux first to adjacent rod without cladding surface thermocouples. Whether or not this phenomena will occur will be determined from the LOFT transient (blowdown) CHF tests discussed in Section VI.

3.2 Bundles Without Rod External Thermocouple Simulators

For the bundles without rod external thermocouples (ANC3 and ANC4), critical heat fluxes were observed on a variety of rods both in the subcooled region and high quality region. For the ANC3 bundle in the high pressure low quality region, CHF burn marks generally extended completely around the rods and in some cases over several inches axiaily^[2]. Compared to the ANC2 bundle, this also provides evidence that the rod external thermocouple simulators in the bundles tested affected the CHF mechanism.

V. CONCLUSIONS

Critical heat flux tests have been conducted on 25-rod bundles simulating the central region of the LOFT nuclear reactor core-1. The effect of fuel rod cladding surface thermocouples on CHF has been evaluated over the pressure range of 6.9 to 13.8 MPa. The following conclusions have been reached based on the test data:

- (1) In the pressure and quality range where CHF is predicted to occur in the LOFT core during blowdown operation (11 MPa and 30 to 40 percent quality), the magnitudes of CHF between the bundles with and without external thermocouple simulators were about the same. Thus, cladding surface thermocouples on LOFT fuel rods are not expected to influence CHF or time-to-CHF on the LOFT fuel rods under loss-of-coolant conditions.
- (2) An insufficient data base was obtained with the bundles with rod external thermocouple simulators to develop a new CHF correlation that could be used to predict CHF for LOFT under blowdown conditions. This resulted from experimental facility limitations on fluid quality during these tests. Additional tests at lower mass velocities are planned (see Section VI) to expand the data base in the range of 10.3 to 11.7 MPa and aid in developing a correlation for predicting transient CHF.
- (3) Steady state correlations used in codes such as RELAP and FRAP-T to predict CHF for LOFT under blowdown conditions have been compared with the data from the bundles with rod external thermocouple simulators. The B&W-2, Barnett and GE CHF correlations do not satisfactorily predict this data and therefore, are not verified as adequate tools to predict CHF for LOFT blowdowns.

VI. FOLLOW-ON CHF TESTS

Additional CHF testing with the rod bundle with external thermocouple simulators is planned. The purpose of this testing is to extend the range of the data, in the pressure range of 10.3 to 11.7 MPa, to mass velocities as low as 136 kg/s-m² and higher qualities than have been previously tested. This data will aid in verifying the use of various steady state CHF correlations to predict time-to-CHF in LOFT during a blowdown or allow a new or modified correlation to be developed.

Also, transient (blowdown) CHF tests on bundles with and without rod external thermocouple simulators have been conducted. The purpose of these tests was to determine if the external thermocouples attached to the cladding surface of some of the LOF; fuel rods will affect CHF (or time-to-CHF) in the LOFT core under blowdown conditions. The rod bundles used for these tests were similar to those used for the steady state CHF tests. The results of these tests are reported in Reference 20.

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APPENDIX A

TIST CONDITIONS

EXX2 RUN CONDITIONS

RUN	SYSTEM PRESSURE PSIA (PA)	INLET TEMPERATURE F (K)	INLET ENTHALPY BTU/LBM (J/KG)	INLET MASS FLUX LBM/HR-FT2 (KG/S-M2)	BUNDLE AVERAGE HEAT FLUX BTU/HR-FT2 (W/M2)		
	2003 (1.385.07)	600. (589.)	615. (1. 63F 06)	1.51E 06 (2045.)	0.429E 06 (1.35E 06)		
	2003. (1.385 07)	606. (592.)	623. (1.45F 06)	2.01F 06 (2730.)	0.485E 06 (1.53E 06)		
0.	1005 (1 386 07)	610. (594.)	629. (1.46F 06)	2.47E 06 (3352.)	0.544E 06 (1.71E 06)		
10	2005. (1.385 07)	603. (590.)	619. (1.44F 06)	1.44F 06 (1953.)	0.428E 06 (1.35E 06)		
11	2003. (1.385 07)	581. (578.)	588, (1.37E 06)	2.01E 06 (2720.)	0.556E 06 (1.75E 06)		
12.	2010- (1-395 07)	587. (582.)	597. (1.39E 06)	2.47E 06 13356.1	0.603E 06 (1.90E 06)		
16.	2001, (1.38F G7)	543. (557.)	539. (1.25E 06)	2.56E 06 (3475.)	0.804E 06 12.53E 001		
18.	2005, (1.38E 07)	565. (569.)	567. 11.32E 061	2.46E 06 (3337.)	0.702E 06 (2.21E 06)		
19.	2005, (1.38F 07)	548. (560.)	545. (1.27E 06)	2.50E 06 (3397.)	0.738E 06 (2.33E 06)		
20.	1997. (1.38E 07)	547. (559.)	544. (1.26E U6)	1.99E 06 (2705.)	0.665E 06 (2.10E 06)		
21.	1995. (1.38E 07)	545. (558.)	541. (1.26E 06)	1.51E 06 (2042.)	0.562E 06 (1.77E 06)		
22.	1500. 11.03E 07)	553. (562.)	552. (1.28E 06)	1.01E 06 (1367.)	0.474E 06 (1.49E 06)		
23.	1500. (1.03E 07)	550. (561.)	549. (1.28E 06)	1.51E 06 (2052.)	0.551E 06 (1.74E 06)		
24.	1515. (1.04E 07)	556. (564.)	556. (1.29E 06)	1.99E 06 (2698.)	0.5868 06 (1.858 06)		
25.	1515. (1.04E 07)	552. (562.)	551. (1.28E 06)	2.50E 06 (3386.)	0.654E 06 (2.06E 06)		
20.	1500. (1.03E 07)	515. (542.)	506. (1.18E 06)	9.85E 05 (1336.)	0.546E 06 (1.72E 06)		
27.	1500. (1.03E 07)	519. (544.)	511. (1.19E 06)	1.57E 06 (2130.)	0.627E 06 (1.98E 06)		
28.	1500. (1.03E 07)	530. (550.)	523. (1.22E 06)	2.01E 06 (2732.)	0.656E 06 (2.07E 06)		
29.	1500. (1.03E 07)	537. (553.)	532. (1.24E 06)	2.48E 06 (3368.)	0.704E 06 (2.22E 06)		
30.	1500. (1.03E 07)	518. (543.)	509. (1.18E 06)	2.44E 06 (3307.)	0.767E 06 12.42E 06)		
31.	1515. (1.04E 07)	493. (529.)	479. (1.11E 06)	1.48E 06 (2008.)	0.686E 06 12.16E 06)		
32.	1500. (1.03E 07)	497. (531.)	484. (1.13E 06)	1.99E 06 (2694.)	0.768E 06 12.42E 061		
33.	1515. (1.04E 07)	501. (534.)	489. (1.14E 06)	2.46E 06 (3341.)	0.828E 06 (2.61E 06)		
34.	1500. (1.03E 07)	467. (515.)	449. (1.05E 06)	9.86E 05 (1337.)	0.607E 06 (1.91E 06)		
35.	1505. (1.04E 07)	467. (515.)	450. (1.05E 06)	1.50E 06 (2037.)	0.744E 06 (2.35E 06)		
38.	2000. (1.38E 07)	562. (568.)	563. (1.31E 06)	1.02E 06 (1386.)	0.454E 06 (1.43E 06)		
43.	2005. (1.38E 07)	483. (524.)	468. (1.09E 06)	9.83E 05 (1333.)	0.527E 06 [1.66E 06]		
46.	2000. (1.38E 07)	513. (540.)	502. (1.17E 06)	2.01E 06 (2728.)	0.751E 06 12.37E 061		
47.	2005. (1.38E 07)	509. (538.)	498. (1.16E 06)	1.53E 06 (2073.)	0.613E 06 (1.93E 06)		
48.	2005. (1.38E 07)	487. (526.)	473. (1.10E 06)	1.50E 06 (2039.)	0.659E 06 (2.08E 06)		
51.	1995. (1.38E 07)	432. (495.)	411. (9.57E 05)	1.02E 06 (1379.)	0.575E 06 (1.81E 06)		

ANCI RUN CONDITIONS

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RUN	SYSTEM PRESSURE PSTA (PA)	INLET TEMPERATURE F (K)	INLET INLET ENTHALPY MASS FLUX BTU/LBM LBM/HR-FT2 (J/KG) (KG/S-M2) 625. (1.45E 06) 1.93E 06 (262 1.93E 06 (262		BUNDLE AVERAGE HEAT FLUX BTU/HR-FT2 (W/M2)		
				$(\alpha,\beta) = (\alpha,\beta) = (\alpha,\beta$	we say us the definition as an an an end of the definition of the two the two the two the two two the two		
14.	1995. (1.38E 07)	607. (593.)	625. (1.45E 06)	1.93E 06 (2623.)	0.472E 06 (1.49E 06)		
15.	2005. (1.38E 07) 2000. (1.38E 07)	593. (585.) 609. (594.)	605. (1.41E 06) 628. (1.46E 06)	1.48E 06 (2007.) 2.45E 06 (3329.)	0.434E 06 (1.37E 06) 0.528E 06 (1.66E 06)		

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ANC2 RUN CONDITIONS

RUN	SYSTEM PRESSURE PSTA	SYSTEM INLET PRESSURE TEMPERATURE PSTA F (PA) (K1		INLET MASS FLUX LBM/HR-FT2	BUNDLE AVERAGE HEAT FLUX BTU/HR-FT2
10.00	(PA)	(K)	137857	(KG/S-M2)	1 #/ 72 /
27.	2005. (1.38E 07)	530, (550,)	523. (1.22E 06)	2.03E 06 (2755.)	0.742E 06 (3.34E 06)
32.	2005. (1.38E 07)	493. (529.)	479. (1.11E 06)	2.03E 06 (2758.)	0.817E 06 (2.58E 06)
37.	2005. (1.38F 07)	466. [514.]	448. 11.04F 06)	2.018 06 (2728.)	0.887E 06 12.80E 061
44 .	2005. (1.38E 07)	436. 1498.)	416. 19.67E 05)	2.03E 06 (2754.)	0.960E 06 (3.03E 06)
56.	2000. (1.38E 07)	504. (535.)	492. (1.14E 06)	1.49E 06 (2025.)	0.648E 06 12.04E 061
61.	2000. (1.38E 07)	471. (517.)	455. (1.06E 06)	1.49E 06 42015.1	0.715E 06 (2.25E 06)
66.	2005. (1.38E 07)	435. (497.)	415. (9.658 05)	1.51E 06 (2054.)	0.788E 06 (2.48E 06)
73.	1995. (1.38E 07)	398. (477.)	375. (8.72E 05)	1.51E 06 (204).	0.840E 06 (2.65E 06)
78.	2005. (1.38E 07)	356. (453.)	330. (7.69E 05)	1.51E 06 (2050.)	0.938E 06 (2.96E 06)
80.	2000. (1.385 07)	346. (448.)	321. (7.46E 05)	1.49E 06 (2020.)	0.954E 06 (3.01E 06)
82.	2000. (1.38E 07)	403. (479.)	380. (8.85E 05)	2.00E 06 (2719.)	0.105E 07 (3.31E 06)
83.	1995. (1.38E 07)	383. (468.)	359. (8.34E 05)	1.99E 06 (2694.)	0.108E 07 (3.42E 06)
105.	2005. (1.386 07)	549. (560.)	546. 11.27E 061	2.54E 06 13446.)	0.788F 06 (2.48E 06)
111.	2005. (1.38E 07)	519. (543.)	509. (1.18E 06)	2.52E 06 (3412.)	0.859E 06 (2.71E 06)
116.	2005. (1.38E 07)	487. (526.)	473. (1.10E 06)	2.51E 06 (3398.)	0.954E 06 (3.01E 06)
123.	2000. (1.38E 07)	454. (508.)	436. (1.01E 06)	2.518 06 (3404.)	0.106E 07 (3.33E 06)
128.	2005. (1.38E 07)	410. (483.)	387. (9.01E 05)	2.53E 06 (3428.)	0.118E 07 (3.72E 06)
150.	2000. (1.38E 07)	443. (501.)	423. 19.84E 051	1.03E 06 (1394.)	0.569E 06 (1.79E 06)
155.	2000. (1.38E 07)	405. (480.)	383. (8.90E 05)	1.02E 06 (1390.)	0.618E 06 (1.95E 06)
162.	2000. (1.38E 07)	366. (459.)	341. 17.94E 051	1.01F 06 (1374.)	0.664E 06 12.09E 061
167.	2000. (1-38E 07)	330. (439.)	304. 17.07E 051	1.011 10 (1370.)	0.694E 06 (2.19E 06)
172.	2000. (1.38E 07)	293. (418.)	266. (6.19E 05)	9.938 95 (1347.)	0.725E 06 12.29E 061
183.	2005. (1.385 07)	382. 1468.1	358. (8.33E 05)	7.58E 05 (1028.)	0.562E 06 (1.77E 06)
190.	2000. (1.388 07)	341. (445.)	315. (7.32E 05)	7.47E 05 (1013.)	0.575E 06 (1.81E 06)
195.	2000. (1.38E 07)	307. (426.)	280. (5.51E 05)	7.57E 05 (1026.)	0.599E 06 (1.89E 06)
201.	2000. (1.38E 07)	266. (403.)	238. (5.54E Q5)	7.46E 05 (1012.)	0.629E 06 (1.98E 06)
203.	2005. (1.38E 07)	260. 1400.1	233. (5.41E 05)	7.44E 05 (1009.)	0.638E 06 (2.01E 06)
204.	2005. (1.38E 07)	379. (466.)	354. (8.25E 05)	7.52E 05 (1020.)	0.568E 06 (1.79E 06)
226.	2005. (1.38E 07)	379. (466.)	354. (8.25E Q5)	7.528 05 (1020.)	0.548E 06 (1.73E 06)
235.	2000. (1.38E 07)	475. (519.)	459. (1.07E 06)	1.25E 06 (1695.)	0.601E 06 (1.89E 06)
240.	2000. (1.38E 07)	440. (500.)	420. 19.78E 051	1.25E 06 (1702.)	0.654E 06 (2.06E 06)
247.	2000. (1.38E 07)	405. 1480.1	383. (8.90E 05)	1.238 06 (1675.)	0.698E 06 (2.20E 06)
252.	1995. (1.38E 07)	371. (462.)	347. (8.06E 05)	1.23E 06 (1673.)	0.745E 06 (2.35E 06)
257.	2000. (1.38E 07)	327. 1437.1	301, (7.00E 05)	1.21E 06 (1648.)	0.794E 06 (2.50E 06)
259.	2000. (1.386 07)	315. (430.)	288. (6.70E 05)	1.22E 06 (1657.)	0.802E 06 (2.53E 06)
321.	2010. (1.39E 07)	482. (523.)	466. (1.08E 06)	1.02E 06 (1383.)	0.542E 06 (1.71E 06)
334.	2005. 01.38E 07)	380. (467.)	356. (8.28E 05)	7.52E 05 (1020.)	0.557E 06 (1.76E 06)
278.	1000. (6.89E 06)	516. 1542.1	507. (1.18E 06)	1.02E 06 (1387.)	0.563E 06 (1.77E 06)
279.	1000. (6.89E 06)	496. (531.)	483. (1.12E 06)	1.02E 06 (1389.)	0.593E 06 (1.87E 06)
280.	1000. (6.89E 06)	458. (510.)	439. (1.02E 06)	1.00E 06 (1360.)	0.645E 06 (2.04E 06)
281.	1000. (6.89E 06)	416. (586.)	393. (9.13E 05)	1.01E 06 (1364.)	0.737E 06 12.32E 061
282.	1000. (6.8°E 06)	371. (462.)	345. (8.03E 05)	9.96E 05 (1351.)	0.808E 06 12.55E 06)
283.	1000. 16.89E 06)	337. (442.)	309. (7.19E 05)	1.01E 06 (1375.)	0.839E 06 (2.64E 06)
284.	1000. (6.89E 06)	318. (432.)	289. 16.73E 051	9.51E 05 (1290.)	0.848E 06 12.67E 061
285.	1500. (1.03E 07)	559. (566.)	560. (1.30E 06)	1.05E 06 (1420.)	0.454E 06 (1.43E 06)
286.	1505. (1.04E 07)	562. (567.)	563. (1.31E 06)	1.048 06 (1417.)	0.453E 06 (1.43E 06)
287.	1505. (1.04E 07)	534. 1552.1	528. (1.23E 06)	1.038 06 (1401.)	0.500E 06 (1.58E 06)
288.	1495. (1.03E 07)	493. (529.)	479. (1.11E 06)	1.03E 06 (1394.)	0.574E 06 (1.81E 06)
289.	1500. (1.03E 07)	460. (511.)	442. (1.03E 06)	1.02E 06 (1378.)	0.633E 06 12.00E 06)
290.	1505, 11.04E 07	411. (483.)	388. 19.025 051	1.01E 06 (1372.)	0.702E 06 (2.21E 06)

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ANCZ RUN CONDITIONS

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RUN	SYSTEM PRESSURE PSIA (PA)	INLET TEMPERATURE F (K)	INLET ENTHALPY BTU/LBM (J/KG)	INLET MASS FLUX LBM/HR-FT2 (KG/S-M2)	BUNDLE AVERAGE HEAT FLUX BTU/HR-FT2 (W/M2)		
291.	1510. (1.04E 07)	359. (455.)	333. 17.74E 05)	9,968 05 (1351.)	0.767E 06 (2.42E 00)		
292.	1495. (1.03E 07)	314. (430.)	287. 16.67E 051	9.62E 05 (1304.)	0.793E 06 12.50E 061		
293.	1005. (6.93E 06)	518. (543.)	509. (1.18E 06)	2.01E 05 (2728.)	0.703E 06 (2.22E 06)		
294.	1005. (6.93E 06)	493. (529.)	479. (1.11E 06)	1.98E 06 (2681.)	0.763E 06 (2.40E 06)		
295.	1000. (6.89E 06)	461. (511.)	442. (1.03E 06)	2.00E 06 (2707.)	0.853E 06 (2.69E 06)		
296.	1000. (6.89E 06)	420. (489.)	398. 19.25E 05)	1.99E 06 (2693.)	0.936E 06 (2.95E 06)		
297.	1005. (6.93E 06)	380. (466.)	354. (8.24E 05)	1.98E 06 (2689.)	0.104E 07 (3.27E 06)		
298.	1495. (1.03E 07)	555. (564.)	555. (1.29E 06)	2.03E 06 (2759.)	0.556E 06 (1.75E 06)		
299.	1505. (1.04E 07)	532. (551.)	525. (1.22E 06)	2.01E 06 (2730.)	0.642E 06 (2.02E 06)		
300.	1510. (1.04E 07)	499. (533.)	487. (1.13E 06)	2.02E 06 (2734.)	0.764E 06 (2.41E 06)		
301.	1500. (1.03E 07)	461. (512.)	443. (1.03E 06)	2.02E 06 (2745.)	0.8398 06 12.648 061		
302.	1505. (1.04E 07)	419. (488.)	396. 19.22E 05)	1.968 06 (2665.)	0.936E 06 (2.95E 06)		
311.	1005. (6.938 06)	521. (545.)	512. (1.19E 06)	5.40E 05 (732.)	0.448E 06 (1.41E 06)		
312.	995. 16.86E 06)	444. (502.)	423. 19.85E 051	5.43E 05 (736.)	0.513E 06 (1.62E 06)		
313.	1000. (6.89E 06)	344. 1447.)	317. (7.37E 05)	5.08E 05 (689.)	0.576E 06 (1.82E 06)		
314.	1495. (1.03E 07)	543. (557.)	539. (1.25E 06)	5.258 05 (712.)	0.373E 06 (1.18E 06)		
315.	1505. (1.04E 07)	443. (501.)	423. (9.83E 05)	5.40E 05 (732.)	0.479E 06 (1.51E 06)		
316.	1500. (1.03E 07)	333. (440.)	306. 17.12E 051	5.00E 05 (678.)	0.565E 06 (1.78E 06)		
317.	1510. (1.04E 07)	246. (392.)	217. (5.05E 05)	5.00E 05 (678.)	0.621E 06 (1.96E 06)		
318.	1005. (6.938 06)	246. (392.)	216. (5.03E 05)	4.99E 05 (677.)	0.663E 06 (2.09E 06)		
319.	1505. (1.04E 07)	246. (392.)	217. 15.05E 051	5.22E 05 (708.)	0.630E 06 (1.99E 06)		
320.	1495. (1.03E 07)	277. (409.)	249. (5.79E 05)	5.13E 05 (696.)	0.627E 06 (1.98E 06)		

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ANCE RUN CONDITIONS

RUN	SYSTEM PRESSURE PSIA (PA)	SYSTEM INLET INL PRESSURE TEMPERATURE ENTH PSTA F BTU (PA) (K) (J/)		INLET MASS FLUX LBM/HR-FT2 (KG/S-M2)	BUNDLE AVERAGE MEAT FLUX BTU/HR-FT2 (W/M2)
25.	2000. (1.385 07)	609. (593.)	627. (1.46E 06)	2.07E 06 (2808.)	0.470E 06 (1.48E CS)
28.	2005. (1.38E 07)	576. (575.)	581. (1.35E 06)	2.046 06 (2770.)	0.579E 06 (1.83E 06)
38.	2005. (1.38E 07)	596. (587.)	609. 11.42E 06)	1.28E 06 (1740.)	0.4158 06 (1.318 06)
42.	2000. (1.38E 07)	528. (549.)	521. (1.21E 06)	1.36E 06-(1841.)	0.582E 06 (1.83E 06)
45.	2015. (1.39E 07)	431. (495.)	410. 19.54E 051	1.24E 06 (1686.)	0.744E 06 12.35E 061
48.	2000. (1.38E 07)	364. (458.)	339. (7.89E 05)	1.21E 06 (1646.)	0.835E 06 (2.63E 06)
51.	2000. (1.38E 07)	341. (445.)	315. (7.34E 05)	1.18E 06 (1599.)	0.859E 06 12.71E 061
57.	2000. (1.38E 07)	500. (533.)	487. (1.13E 06)	2.02E 06 (2745.)	0.812E 06 (2.56E 06)
62.	2000. (1.38E 07)	447. (504.)	428. (9.95E 05)	2.02E 06 (2739.)	0.995E 06 (3.14E 06)
67.	2000. (1.38E 07)	417. (487.)	395. (9.20E 05)	2.03E 06 12754.)	0.108E 07 (3.40E 06)
70.	2000. (1.38E 07)	396. (475.)	373. (8.68E 05)	2.01E 06 (2732.)	0.110E 07 (3.48E 06)
80.	2005. (1.38E 07)	545. (558.)	541. (1.26E 06)	1.50E 06 (2031.)	0.555E 06 (1.75E 06)
99.	2000. (1.38E 07)	603. (590.)	618. (1.44E 06)	2.54E 06 (3439.)	0.533E 06 (1.68E 06)
100.	2000. (1.38E 07)	577. 1576.1	582. (1.35E 06)	2.51E 06 (3405.)	6.653E 06 (2.06E 06)
107.	2000. (1.38E 07)	495. (530.)	481. (1.12E 06)	2.53E 06 (3428.)	0.991E 06 (3.12E 06)

ANC4 RUN CONDITIONS

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RUN	SYSTEM PRESSURE PSIA (PA)	SYSTEM INLET PRESSURE TEMPERATURE PSIA F (PA) (K)		INLET MASS FLUX LBM/HR-FT2 (KG/S-M2)	BUNDLE AVERAGE HEAT FLUX BTU/HR-FT2 (W/M2)			
		830 / 650 V	5.24 (1) 22E 061	4.91F 05 (666.)	0.422E 06 (1.33E 06)			
10.	1005. (6.938 06)	550. (550.)	431 11.00F 061	5.016 05 (679.)	0.482E 06 (1.52E 06)			
11.	1000. 16.895 067	530 1550.1	526. (1.22F 06)	5.16E 05 (700.)	0.379E 06 (1.19E 06)			
12.	1515. 11.040 077	541 (556.)	537. (1.25E 06)	5.10E 05 (692.)	0.363E 06 (1.14E 06)			
13+	1500. (1.036 07)	636 (696.)	413, 19,61E 05)	4.99E 05 (677.)	0.449E 06 (1.42E 06)			
1.4.	1515. 11.046 071	336. 1442.1	309, 17,19E 05)	5.00E 05 (678.)	0.547E 06 11.72E 061			
12.	1000 14 895 061	358. (454.)	332. (7.71E 05)	4.96E 05 (673.)	0.558E 06 (1.76E 06)			
21	1000. (0.04E 00)	561, (567.)	562. (1.31E 06)	9.97E 05 (1352.)	0.454E 06 (1.43E 06)			
22	1500 (1.035 07)	537. (554.)	533. (1.24E 06)	1.00E 06 (1363.)	0.493E 06 (1.55E 06)			
22.	1000. (6.89E 06)	513. (540.)	503. (1.17E 06)	1.01E 06 (1364.)	0.565E 06 (1.78E 06)			
24.	1515, (1-04F 07)	494. (530.)	481. (1.12E 06)	1.01E 06 (1374.)	0.550E 06 (1.73E 06)			
25.	1500, (1,03F 07)	468. (515.)	451. (1.05E 06)	9.83E 05 (1333.)	0.590E 06 (1.86E 06)			
26.	1000, (6,89F 06)	456. (509.)	437. (1.02E 06)	9.82E 05 (1332.)	0.651E 06 (2.05E 06)			
27.	1500, (1.03E 07)	415. (486.)	392. (9.12E 05)	1.02E 06 (1383.)	0.6752 06 (2.132 06)			
28.	1500, 11,03F 071	560. 1566.)	561. (1.30E 06)	1.98E 06 (2679.)	0.561E 06 (1.77E 06)			
29.	1500. (1.03E 07)	535. (553.)	530. (1.23E 06)	2.04E 06 (2765.)	0.653E 06 (2.06E 06)			
89.	1000, 16.89E 061	347. (448.)	320. (7.44E 05)	1.08E 06 (1471.)	0.799E 06 12.52E 061			
90.	1500. (1.03E 07)	322. (434.)	295. (6.86E 05)	1.00E 06 (1357.)	0.751E 06 (2.37E 06)			
91.	1500. (1.03E 07)	360. (456.)	335. (7.79E 05)	1.05E 06 (1421.)	0.758E 06 (2.39E 06)			
92.	1000. (6.89E 06)	379. (466.)	354. (8.23E 05)	1.05E 06 (1424.)	0.783E 06 (2.47E 06)			
93.	1000. 16.89E 06)	417. (487.)	395. (9.18E 05)	1.05E 06 (1428.)	0.718E 06 12.26E 06)			
95.	2000. (1.38E 07)	425. (492.)	404. (9.40E 05)	1.25E 06 (1692.)	0.684E 06 (2.16E 06)			
96.	1005. (6.93E 06)	455. (508.)	436. (1.01E 06)	2.04E 06 (2773.)	0.851E 06 (2.68E 06)			
97.	1005. (6.93E 06)	501. (534.)	489. (1.14E 06)	1.988 06 (2690.)	0.737E 06 12.32E 061			
98.	2000. (1.38E 07)	501. (534.)	489. (1.14E 06)	2.04E 06 (2761.1	0.800E 06 (2.52E 06)			
100.	2000. (1.38E 07)	509. (538.)	498. 11.16E 061	1.98E 06 (2689.)	0.7502 06 (2.362 06)			
102.	1000. (6.89E 06)	515. (542.)	506. (1.18E 06)	1.998 06 (2698.)	0.691E 06 (2.18E 06)			
104.	1500. (1.03E 07)	267. 1404.1	239. (5.55E 05)	4.98E 05 (675.)	0.0026 00 (1.916 00)			

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APPENDIX B

LOCAL SUBCHANNEL CONDITIONS FOR ALL CHF DATA POINTS

LOCAL SUBCHANNEL CHE CONDITIONS (SI UNITS) EXX2 DATA

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RUN ND.	CHF ROD NO.	CHF CHANNEL	PRESSURE (KPA) X 10-4	MASS VEL: (KG/5-M2) X 10-3	EQUILIBRIUM	PRE LOFT2	U 1 6 1 (W/M2) 66w-2	E D X 10-6 BARNETT	с н ғ GE	MEASURED CHF (W/M2) X 10-6 HFLUX
22	12-2	30	1.034	1.391	0.195	1.99	1.44	1.20	1.90	1.50
23	21-2	35	1.034	1.818	0.168	1.94	1.46	1.34	1.99	1.77
26	21-2	85	1.045	2.435	0.122	1.99	1.70	1.45	2.14	1.88
25	21-2	35	1.045	3.032	0.090	2.08	1.94	1.57	2.24	2.10
26.1	21-2	35	1.024	1.124	0.243	1.91	1.27	1.20	1.76	1.75
26.2	12-2	30	1.034	1.359	0.170	2.13	1.74	1.33	1.99	1.73
27	21-2	35	1.034	1.870	0.128	2.16	1.90	1.58	2.12	2.01
28	21-2	35	1.034	2.408	0.092	2.23	2.12	1.70	2.23	2.11
22.1	21-2	15	1.034	2.185	0.074	2.24	2.18	1.73	2.29	2.26
23.2	12-2	30	1.034	3.379	0.042	2.46	2.92	1.74	2.39	2.55
20	21-2	35	1.034	2.888	0.056	2.40	2.40	1.91	2.35	2.46
31	21-2	35	1.045	1.729	0.119	2.23	2.07	1.66	2.15	2.20
32	8-2	27	1.034	2.552	7.041	2.56	2.71	1.85	2.39	2.56
22	8-2	27	1.045	3,193	0.009	2.73	3.03	2.03	2.49	2.76
36-1	8-2	27	1.034	1.281	0.137	2.31	2.15	1.51	2.09	2.02
34.2	20-0	26	1.034	1.192	0.245	1.85	1.10	0.80	1.75	1.90
36	21-2	35	1.038	1.717	0.093	2.38	2.39	1.83	2.23	2.39
52	21-2	30	1.045	1.144	0.160	2.24	2.00	1.57	2.02	2.08
7	8-2	27	1.382	1.987	0.109	1.60	1.80	1.25	2.18	1.43
8	8-2	27	1.382	2.547	0.092	1.55	1.96	1.22	2.23	1.61
	8-2	27	1.376	3.258	0.089	1.44	2.02	1.14	2.24	1.61
10.1	21-2	35	1.382	1.810	0.154	1.40	1.43	1.19	2.04	1.37
10.2	25-2	35	1.392	1.810	0.154	1.40	1.43	1.19	2.04	1.38
11	8-2	27	1.382	2.623	0.049	1.86	2.37	1.48	2.37	1.85
12	R = 2	27	1.386	3.226	0.037	1.88	2.52	2.50	2.40	2.01
16	8-2	27	1.379	3.406	-0.030	2.46	3.22	1.98	2.62	2.68
1.0	8-2	27	1.382	3.218	800.0	2.13	2.82	1.71	2.50	2.34
19	8-2	27	1.382	3.338	-0.033	2.48	3.24	2.00	2.63	2.46
20	8-2	27	1.376	2.620	0.0	2.21	2.33	1.79	2.52	2.21
21	18-0	34	1.376	1.860	0.116	1.61	1.70	1.00	2.16	1.69
39.1	21-2	35	1.379	1.276	0.170	1.54	1.32	1.25	1.99	1.46
38.2	24-0	36	1.379	1.229	0.208	1.39	1.04	1.11	1.87	1.42
41	8-1	27	1.382	1.262	0.042	2.06	2.27	1.51	2.39	1.75
46	8-1	27	1.379	2.572	-0.029	2.39	3.09	1.96	2.61	2.50
47	1-6	27	1.382	1.945	-0.015	2.27	2.85	1.87	2.68	2.03
48	8-1	27	1.382	1.926	-0.036	2.37	3.03	1.97	2.64	2.19
61	8-1	27	1.376	1.323	-0.055	2.39	3.04	1.99	2.69	1.92

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LOCAL SUBCHANNEL CHE CONDITIONS ISL UNITSI ANCI DATA

			PRESSURE	MASS VEL.		PRE	DICT	ED	CHF.	MEASURED CHF
ALIN: 107	CHF	CHE	(KPA) X 10-4	(KG/S-M2)	CHALITY	LOFT2	(W/MZ) BEW-2	K LC-6	T GE	HELUX
					And the set of the set of		and the second second			
14	8-2	27	1.376	2.548	0.101	1.52	1.88	1.19	2.20	1.57
15	8-2	27	1.392	1.951	0.094	1.70	1.92	1.33	2.23	1.44
17	8-2	27	1.379	3.229	0.083	1.49	2.08	1.19	2.26	1.75

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LOCAL SUBCHANNEL CHE CONDITIONS (SI UNITS) ANC2 DATA

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	CHE	CHE	PRESSURE (KPA)	MASS VEL. (KG/S-M2)	EQUILIBRIUM	PRE	0 1 C T (W/M2)	E D X 10-6	Снғ	(W/M2) X 10-6
RUN NO.	ROD NO.	CHANNEL	X 10-4	X 10-3	QUALITY.	LOFT2	8£₩-2	BARNETT	GE	HFLUX
270	23-2	31	0.4.90	1.433	0.309	2.12	0.29	0.60	1.55	1.83
270	23-2	21	0.689	1 100	0.295	2.78	0.57	0.67	1.59	1.95
220	23-2	21	0.629	1 204	0.293	2.62	0.99	1. 76	1 63	2.12
201	23-2	31	0.689	1 260	0.276	2 68	1.11	0.20	1.45	2.42
201	20.1	31	0.007	0.042	0.210	2.40	1.20	0.00	1 30	2.57
202+1	20-1	32	0.609	0.942	0.361	2.30	1.20	0.05	1.30	2.66
202 + 2	21-2 21-2	32	0. (00.9	0.742	0.200	2 61	1.20	1 12	1 4 3	2.66
202.02	21-2	35	0.009	1 362	0.303	2.01	1.49	0.83	1.57	2.66
292.44	22-2	20	0.639	1.200	0.260	2.40	1.51	0.07	1.70	2.66
206+2	23-2	36	0.690	0.00.0	0.200	2.57	1.77	1.04	1.63	2.76
202	20-1	35	0.690	0.071	0.341	2 66	1 82	0.08	1.45	2.66
2.046	20-1	33	1.034	1 271	0.251	1 78	1.00	1.01	1.73	1.49
202	21-2	35	1.034	1.268	0.256	1.75	0.95	0.99	1.71	1.49
200+1	21-7	32	1.038	1.260	0.263	1.66	0.76	0.70	1.70	1.49
200.2	21-2	20	1.030	1.070	0.241	1.70	0.94	0.72	1.76	1.69
200.3	22-2	21	1 028	1.478	0.261	1.70	0.84	0.72	1.76	1.42
200.44	21-2	24	1.038	1 226	0.236	1.99	1.21	1.10	1.79	1.65
200 1	20-1	35	1.031	1.100	0.259	1,83	1.08	1.04	1.71	1.79
200.1	20-1	32	1.031	1 073	0.263	1.86	1.09	0.80	1.60	1.70
200+2	20-2	34	1.031	1 245	0.220	1.46	1.26	0.85	1.83	1.79
200.0	20-5	36	1.031	1.203	0.200	2.03	1.50	1.23	1.96	1.79
200.4	21-2	3.6	1.031	1.203	0.209	2.03	1.50	1-23	1.86	1.89
200.1	20-1	35	1.034	1.112	0.251	1,88	1.19	1.08	1.73	1.97
234.2	21-2	35	1.036	1.173	0.194	2.11	1.67	1.30	1.91	2.08
232.7	26-2	33	1-034	5.313	0.227	1.88	1.15	0.82	1.81	1,98
240.1	20-1	35	1.038	1.093	0.215	2.05	2.11	1.23	1.84	2.19
200.2	21-1	36	1.038	1.093	0.215	2.05	2.11	1.23	1.84	2.31
240.3	21-2	35	1.038	1,157	0.150	2.29	1.55	1.48	2.05	2.31
291.1	18-1	3.6	1.041	0.929	0.256	1.96	1.31	0.86	1.71	2.34
291.2	21-1	35	1.041	1.055	0.179	2.21	1.92	1.38	1.96	2.53
292	18-1	34	1.031	0.955	0.133	2.45	1.70	1.10	2.08	2.42
293	21-2	35	0.693	2.426	0.188	2.35	0.43	1.01	1.93	2.32
294.1	21-2	35	0.693	2.328	0.168	2.54	0.90	1.20	1.99	2.51
294.2	23-2	31	0.693	2.655	0.160	2.47	0.64	0.81	2.02	2.51
294.3	24-2	31	0.693	2.655	0.160	2.47	0.64	0.81	2.02	2.39
295.1	21-2	35	0.689	2.298	0.145	2.73	1.34	1.39	2.06	2.81
295.2	23-2	31	0.689	2.609	0.136	2.68	2.13	0.96	2.09	2.81
295.3	24-2	31	0.689	2.609	0.136	2.68	1.13	0.96	2.09	2.67
296	24-2	31	0.689	2.524	0.095	3.01	1.93	1.20	2.22	2.93
297.1	8-3	26	0.693	2.210	0.082	3.16	2.43	1.33	2.26	3.42
297.2	21-2	35	0.693	2.164	0.073	3.22	2.75	1.08	2.29	3.42
2.98	21-2	35	1.031	2.545	0.112	2.06	1.77	1.36	2.17	1.83
299.1	20-3	26	1.038	2.559	0.103	2.11	1.81	1.00	2.20	2.00
299.2	21-2	35	1.038	2.450	0.096	2.19	2.03	1.49	2.22	2.11
299.3	24-2	31	1.038	2.740	0.089	2.17	1.93	1.03	2.24	2.01
300.1	8-2	27	1.041	2.559	0.046	2.51	2.65	1.83	2.38	2.52
300.2	8 - 3	26	1.041	2.487	0.081	2+28	2.11	1.11	2+27	2.52
300.3	21-2	35	1.041	2.3B7	0.073	2.35	2.35	1,66	5 * 5 8	2.52
300.4	21-3	26	1.041	2.487	0.081	2.28	2.11	1.11	2.27	2.52
301	8-3	26	1.034	2.476	0.029	2,64	2.78	1.60	2.43	2.16

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LOCAL SUBCHANNEL CHE CONDITIONS (SI UNITS) ANG2 DATA

RUN NO.	CHF ROD NO.	CHF CHANNEL	PRESSURE (KPA) X 10-4	MASS VEL. (KG/5-M2) X 10-3	EQUILIBRIUM QUALITY	PRE	D 1 C T (W/M2) B&W-2	E 0 X 10-6 BARNETT	С Н F GE	MEASURED CHF (W/M2) X 10-6 HFLUX
										3.00
302-1	8-5	21	1.038	2+640	-0.051	3.16	3.95	2.33	2.08	3.08
302.2	9-3	20	1.038	2+414	-0.011	2+89	3.24	1.24	2.00	3.08
302.3	25-1	36	1.058	2.206	0.020	2.11	3.07	2.04	2.40	3.07
911	24-1	31	0.695	0.805	0.521	1+23	0.13	0.34	0.80	1.40
312	24-1	31	0.686	0.154	0.497	1.19	0.11	0.33	0.84	1.00
313	24-1	31	0.084	0.640	0.965	2.03	1.42	0.57	1.00	1.80
314+1	21-2	35	1+031	0.032	0.428	4.40	0.50	0.70	1.30	1.23
*14.2	21-3	20	1.031	0.040	0.433	1. 30	0.33	0.55	1+10	1.23
312.1	20-1	30	1.038	0.601	0.430	1.41	0.01	0.71	0.91	1.47
315.2	21-1	35	1+138	0+601	0 + 4 3 0	1.41	0.01	0.71	0.91	1.50
316	21-1	35	1.034	0.340	0.422	1.58	0.00	0.10	0.91	1.00
317	21-1	35	1.041	0.003	0.331	1.71	1.40	0.41	1+77	2.03
318.1	20-1	33	9.693	0.523	0.214	1.91	2.20	0.03	1.03	2.00
310.2	21-1	35	0.693	0.223	0.480	1.9/	2+20	0.03	1.03	2.10
318.3	24-1	24	0.693	0.650	0.454	2.10	1.71	0.01	1.20	2.01
319	21-1	35	1.038	0.586	0.311	2.00	1+33	1.03	1.07	2.07
320.1	20-1	35	1.031	0.564	0.392	1.70	1.03	0.83	1.41	1.95
320.2	24-1		1.031	0.089	0.379	1+12	0+72	0.70	1.39	1.90
27.1	8-2	21	1.382	2+077	-0.013	2.10	2.31	1.00	2.20	2.44
21.2	20-0	32	1.332	2.470	0.010	2.10	2.00	2.20	2.41	2.43
32	8-2	21	1+ 382	2 1 1 50	-0.014	2.13	1.00	2.29	2.11	2.67
37	8-2	21	1 + 302	2.038	-0.113	2.92	3+91	2 + 47	2.88	2.92
44.44	8-2	21	1.352	2.011	-0.123	1.10	9.67	2 . 14	3.00	3.10
20.1	8-1	21	1.3/3	1.077	0.0014	6+14	2.01	1.13	2.40	2.13
20.00	18-1	14	1.379	1.095	0.040	1.70	1.03	0.98	2.22	1.98
01-1	18-1	34	1+379	1.009	0.007	1.90	2.00	1.07	2+31	2.18
01+2	8-1	21	1.3/9	1.001	0.122	2 . 32	2.93	1.73	2.00	2.33
00+1	8-2	21	1.302	2.031	-0.122	2.10	2.01	2.44	2.71	2.59
10+2	8-1	21	1.302	1.910	-0.075	6.30	2.29	2+19	2+10	2+39
73.1	8+1	21	1.376	1.408	-0.125	2.18	3+84	2.42	2.92	2.11
13+2	18-1	34	1.376	1.741	-0.030	2.34	2.85	1.30	2.02	2.56
/3.1	8-1	21	1.382	1.934	-0.162	2.91	4.13	2.02	3.03	3.09
78.2	18-1	34	1.382	1.142	-0.058	2.97	3.06	1.44	2.10	2.86
78.3	25-1	36	1.382	1.904	-0.112	2.10	3.09	2.30	2.87	3.08
80.1	8-1	21	1+3/9	1.712	-0+102	2.90	4.13	2.01	5.03	3.14
80.2	18-1	34	1+3/9	1.700	-0.000	2.99	3.04	1.99	2.70	2.91
80.3	25-1	30	1.379	1.009	-0.107	2.07	3.00	2.30	2.81	3.13
22	0-2		1+379	2.000	-0.214	3.32	4.00	2.94	3.11	3.45
83	0-6	21	1.202	2.322	-0.010	3.44	4.04	3.04	3.20	3.27
106	8-2	21	1.302	2.221	-0.010	2+29	3.02	1.85	2.22	2.54
111	0-2	21	1+382	3.319	-0.004	2.14	3.31	2.24	2.12	2.83
110	8-2	2.1	1.382	3 . 300	-0.111	3.09	4+113	2.51	2.87	3.14
123	6-2	61	1.319	3+692	-0.102	3.40	4.47	2.30	3.00	3.49
128.1	8-1	21	1.382	3+107	-0.103	3.00	4.12	3.05	3+10	3.89
128.2	8-2	21	1.382	3+202	-0.223	1.84	2.12	3.35	3.22	3.89
150	18-1	34	1.379	4.102	0.011	1.97	1.97	1.09	2 + 30	1.14
155	18-1	34	1.379	1.108	0.033	2.10	2.25	1.10	2.42	1.88
162	18-1	34	1.379	1.172	-0.003	5.55	2.51	1.27	2.53	2.03
167	8-3	26	1.379	1.390	-0.163	2.69	3.14	1.69	3.04	2.29
172	8-2	27	1.379	1.223	-0.266	2.77	4.61	2.15	3.36	2.39

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LOCAL SUBCHANNEL CHE CURDITIONS (SI UNITS) ANC2 DATA

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RUN NO.	CHF RCD NO.	CHF CHANNEL	PRESSURE (KPA) X 10-4	MASS VEL: (KGZS-M2) X 10-3	EQUILIBRIUM QUALITY	PRE LOFT2	U I C T (w/M2) BCW-2	E D X 10-5 BARNET	СНF TGE	MEASURED CHE (W/M2) X 10-6 HELUX
				And (1997) 1997 1997 1997 1997 1997		3.05	7.04	1.53	2 33	1.84
183	25-1	36	1,382	0.432	0.000	2+04	2+40		2 75	1 80
190	8-3	26	1.379	1.041	-0.071	2 + 30	2.33	1 + 37	2.417	1.04
195	8-3	26	1.379	1.102	-0.128	2.51	3.37	1.03	2.93	1.41
201	0-3	26	1.379	1.074	-0.173	2.57	3.67	1.63	3.0.8	2.07
203	8-3	26	1.382	1.078	-0.173	2.57	3.64	1.62	3.07	2.10
204	25-1	34	1.382	0.975	0.136	1.83	1.49	0.96	2.09	1.86
336	25-1	36	1.382	0.934	0.036	2.11	2.25	1.60	2.41	1.80
220	C 2 - X	27	1.379	1.580	-0.015	2.27	2.80	1.88	2.57	1.98
2 3 3	0-1	27	1 370	1.628	-0.062	2.45	3.19	2.03	2.72	2.15
6.43	5-1	61	1	3 776	-0.101	2.61	3.40	1.56	2.84	2.30
241	8-3	20	1	1	-0.161	2.74	3.71	1-68	2.97	2.46
252	B = 4	26	1+3/0	1.003	0.262	2.94	4.69	2.67	3.32	2-62
257+1	8-2	27	1.379	1.4.24.7	-0.036	2. 94	4.07	1 83	3 13	2-62
257.2	8-3	26	1.379	1.040	-0.199	2.00	4.10	1.03	3.10	2.66
259.1	8-2	27	1.379	1.700	-0.278	3 + 1 2	2+01	3.08	3+40	2 + 0 4
·· 0.2	8-2	26	1.379	1.635	-0.219	2.94	4.29	1+90	3.21	2 + 04
321	25-1	36	1.386	1.219	C.094	1.87	1.88	1.43	2.23	1.78
334	25-1	36	1.382	0.929	0.653	2.06	2+13	1.55	3.03	1.83

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		F.115	PRESSURE	MASS VEL.	ECHIT LEGILLA	PRE	DICT	EC	CHF	MEASURED CHF
	CHF	CHE	INPAL	140/5-02/	CULLIDRIUM	1.05.72	85 8-2	BADNET	r ce	HEILIY
RUN NU.	RED NO.	CHANNEL	X 10-4	x 10-3	WURLIIT	CUPIE	05.8-2	DARTE	i uc	THE LON
							3 4 7	0.74	2.24	
25	23-2	- 31	1.379	2.818	0.116	1 + 32	1.07	0.14	2.10	1.30
28	23-2	31	1.382	2.730	0.069	1.70	2.10	0.95	2.30	1.92
38	21-3	26	1.382	1.755	0.162	1.38	1.29	0.76	2.01	1.37
42.1	21-1	35	1.379	1.637	0.104	1.73	1.83	1.35	2.19	1.93
42.2	23-1	31	1.379	1.722	0.116	1.65	1.65	0.90	2.16	1.93
45.1	8-2	26	1.387	1.591	-0.019	2.27	2.68	1.30	2.58	2.46
43.2	7-2	26	1.389	1.591	-0.019	2.27	2.68	1.30	2.58	2.45
45.3	21-1	35	1.389	1.458	0.030	2.07	2.39	1.66	2.43	2.45
48	8-2	26	1.379	1.512	-0.089	2.54	3.26	1.51	2.80	2.77
51	8-2	26	1.379	1.589	-0.107	2.60	3.39	1.56	2.86	2.85
57	8-2	26	1.379	2.689	-0.030	2.41	2.99	1.36	2.62	2.69
62	8-2	26	1.379	2.731	-0.074	2.70	3.00	1.54	2.75	3.30
67	12-2	30	1.379	2.651	-0.146	3.11	4.21	5.84	2.98	3.59
70	12-2	30	1.379	2.610	-0.182	3.30	4.27	6.05	3.10	3.67
80	12-1	30	1.382	1.877	0.062	1.89	2.19	1.51	2.33	1.85
99	21-2	35	1.378	3.271	0.073	1.58	2.18	1.26	2.29	1.77
100.1	21-2	35	1.379	3.184	0.041	1.87	2.49	1.49	2.39	2.17
100.2	21-3	26	1.379	3.351	0.048	1.79	2.33	0.9	2.37	2.17
107	14-2	31	1.379	3.394	-0.047	2.60	3.26	1.44	2.58	3.27

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LOCAL SUBCHANNEL CHE CUNDITIONS (SI UNITS) ANC3 DATA

LOCAL SUBCHANNEL CHE CONDITIONS (SI UNITS) ANC4 DATA

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RUN NO.	CHF RCD NO.	CHF CHANNEL	PRESSURE (KPA) X 10-4	MASS VEL. (KG/S-M2) X 10-3	EQUILIBRIUM QUALITY	PRF LOFT2	D I C T (W/M2) AGW-2	E D X 10-6 BARNETT	G H F	MEASURED CHF (W/M2) X 10-6 HFLUX
10.1	15-1	31	0.693	0.671	0.597	1.30	0.31	0.29	0.77	1.41
10.2	15-2	31	0.093	0.674	0.547	1.59	0.76	0.40	0.92	1.41
10.2	23-1	31	0.693	0.671	0.597	1.30	0.31	0.29	0.77	1.38
10.5	21-1	35	0.693	0.610	0.600	1.36	0.76	0.28	0.76	1.41
10.5	21-2	35	0.693	0.616	0.543	1.69	1.25	0.45	2.94	1.41
11.1	15-2	31	986.0	0.035	0.489	1.97	1.56	0.55	1.11	1.60
11.2	R-2	27	0.689	0.637	0.412	2.35	2.34	0.86	.35	1.57
12.	15-2	31	1.045	0.681	0.409	1.47	0.51	0.62	1.23	1.26
13.	15-2	31	1.034	0.673	0.421	1.44	0.47	0.59	1.32	1.21
14.	15-2	31	1.045	0.647	0.335	1.83	1.14	0.75	1.59	1.49
15.	23-1	31	1.034	0.617	0.383	1.68	0.90	0.68	1.44	1.79
16.	8-2	27	0.689	0.643	0.356	2.60	2.84	1.04	1.53	1.82
21.	15-2	31	1.045	1.306	0.267	1.65	0.75	0.69	1.68	1.51
22.	15-2	31	1.034	1.292	0.245	1.80	0.99	0.77	1.75	1.64
23.	14-2	31	0.689	1.280	0.326	2.20	0.49	0.61	1.49	1.87
24 .	19-2	31	1.045	1.275	0.195	2.03	1.48	0.92	1.91	1.83
25.	10-2	28	1.034	1.218	0.155	2.26	2.02	1.45	2.03	1.95
26 .	14-2	31	0.689	1.140	0.313	2.38	1.07	0.73	1.54	2.16
27.	14-1	31	1.034	1.205	0.186	2.13	1+65	0.96	1.94	2.23
28.	14-2	31	1.034	2.614	0.133	1.87	1.39	0.84	2.10	1.86
29.	12-2	30	1.034	2.609	0.082	2.26	2.15	1.56	2.26	2.17
89.	15-1	31	0.689	1.196	0.233	2.74	1.68	0.97	1.79	2.66
90.	23-1	31	1.034	1.159	0.079	2.57	2.69	1.27	2+27	2.46
91.1	23-1	31	1.034	1.212	0.135	2.34	2.12	1.11	2.10	2.48
91.2	14-1	31	1.034	1.212	0.135	2.34	2.12	1.11	2.10	2.51
92.	14-1	31	0.689	1.169	0.297	2.44	1.18	0.77	1.59	2.59
93.	14-1	31	0.689	1.216	0.304	2.37	0.94	0.72	1.56	5*38
95.	15-2	31	1.379	1.713	-0.077	2.52	3.19	1.49	2.76	2.28
96.	21-2	35 -	0.693	2.468	0.117	2.86	1.08	1.56	2.15	2.84
97.	21-2	35	0.693	2.438	0.165	2,52	0.82	1.17	2.00	2.46
98.	15-2	31	1.379	2.719	-0.031	2.42	3.01	1.36	2.62	2.66
100.	15-2	31	1.379	2.548	-0.326	2.38	2.95	1.34	2.60	2.50
102.	21-2	35	0.689	2.472	0.174	2.45	0.62	1.09	1.97	2.30
104.	23-1	31	1.034	0.607	0.354	1,81	1.13	0.73	1+53	1.98

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LOCAL SUBCHANNEL CHE CONDITIONS (ENGLISH UNITS) EXX2 CATA

	9-01 X																																						
	IBIU/HK-FIZI		0.476	0.561	0.597	0.666	0.556	0.549	0.639	0.668	0.717	0.808	0.782	0.699	0.811	0.875	0.641	0.603	0.758	0.659	0.453	0.512	0.574	0.435	0.439	0.587	0.637	0.849	141.0	0.179	0.702	0.535	0.463	0.452	0.556	0.793	0.645	0.696	0.608
1 2 3	- Q		0.604	0.632	0.678	0.11.0	0.557	0.630	0.672	C. 708	0.126	0.758	0.744	0.681	651.0	161.0	0.663	0.5555	0.707	0.640	0.691	0.708	C.711	0.646	0.646	0.751	0.763	0.830	0.792	0.833	0.800	0.684	0.630	265.0	0.758	0.829	0.849	0.836	0.855
1	AANETT		0.381	0.424	0.461	0. 498	0.380	0.423	0.501	615.0	645*0	0.551	0. 605	0.528	185-0	0.645	61 ** 0	6.253	0.581	0.498	0.396	0.387	0.363	0.378	0.378	0.471	114.0	0.628	0.542	0.633	0.568	0.317	168.0	0.351	0.512	0-622	665*0	0.626	0.632
	R. W-7 8		0.458	0.462	0.540	710.0	0.402	1:55.0	0.604	0.574	6.693	0.801	0.782	0.657	0.861	0.961	0.681	0.348	0.757	0.634	0.570	0.623	0.640	0.455	0.455	0.753	0.801	1.022	0.894	1+029	0.898	0.540	0.420	0.329	0.720	186.0	0.905	0.962	0.966
A A A	1.0612		0.631	0.616	0.632	0.660	0.606	0.676	0.684	0.708	0.712	0.780	0.763	202*0	0.811	9.867	0.733	0.590	0.756	0.712	0.509	164.0	0.457	0.4444	0.444	165*0	0.598	0.782	0.677	0.786	0.701	0.511	064*0	0.441	0.653	0.759	0.721	0.753	0.757
And the state attent	EQUIL IBRIUM	a section of the sect	0.196	0.168	0.122	0.040	0.243	0.170	0.128	0.092	0.074	0.042	0.056	0.119	1.0.0	0.009	0.137	0.245	0.043	0.160	0.109	0.092	0.089	0*154	0.154	0.049	0.037	-0,030	0.008	-0.033	0.0	0.116	0.170	0.208	0+042	-0.029	-0.015	-0.036	-0.055
MAUU KEL.	(LBM/HR-FI2) x 10-6		1.026	1.341	1.796	2.236	0.829	1.002	1.379	1.776	2.201	2.492	2.130	1.275	1.882	2.355	0.945	0.879	1.266	0.844	1.465	1.952	2.403	1.335	1.335	1.934	2.379	2.512	2.373	2.462	1.932	1.372	0.941	0.906	166.0	1.897	1.434	1.420	0.976
	PRESSURE		1500.	1500.	1515.	1515.	1500.	1500.	1500.	1500.	1500.	1500.	1500.	1515.	1500.	1515.	1500.	1500.	1505.	1515.	2005.	2005.	1995.	2005.	2005.	2005.	2010.	2000.	2005.	2005.	1995.	1995.	2000.	2000.	2005.	2000.	2005.	2005.	.2995.
	CHE		30	35	3.6	35	35	0.0	35	35	35	30	35	35	2.7	27	22	26	35	35	22	27	27	35	35	27	27	27	27	27	27	34	3.6	36	27	27	27	27	27
	CHF WOOL OF	* 100 MC	12-2	21-2	2-12	21-2	21-2	12-2	21-2	21-2	21-2	12-2	21-2	21-2	8-2	8-2	8-2	20-0	21-2	21-2	2-8	8-2	8-2	21-2	25-2	8-2	8-2	8-2	8-2	8-2	8-2	18-0	21-2	0-72	1-8	8-1	2+1	8-1	1-0
		4 LW 1911.	22	23	24	25	20.1	24.2	27	28	1.00	2.62	1.01	11	32	33	34.1	34.2	36	6.5		æ	0	10.1	10.2	11	12	16	1.8	61	20	17	38.1	38.2	43	44	47	8.7	15

LOCAL SUBCHANNEL CHF CONDITIONS (ENGLISH UNITS) ANCI DATA

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	1. 1. 1. 1. 1.			MASS VEL.	CONTRACTOR IN	PRE	DICI	ED	CHF	MEASURED CHF
RUN NO.	ROD NC.	CHANNEL	(PSIA)	X 10-6	QUALITY	LOFTZ	BEN-2	BARNETT	GE	HELUX
		and the set of a set of a line						100 - 100 (10 - 100 - 100)		the real of the last of the last
14	8-2	27	1995.	1.879	0.101	0.481	0.597	0.378	0.699	0.498
15	8-2	27	2005.	1.439	0.094	0.539	0.610	0.422	0.706	0.457
17	8-2	27	2000.	2.381	0.083	0.474	0.659	0.377	0.717	0.555

10 A 10 C 1 A 1

LOCAL SUBCHANNEL CHE CONDITIONS (ENGLISH UNITS) ANC2 DATA 9-01

				MASS VEL.		PRE	0 1 C 1	E 0	C H F	MEASURED CHF
RUN NO.	RDD NO.	CHANNEL	FRESSURE [PSIA]	C C	QUAL . FY	LOF 12	86W-2	BARNETT	99	HFLUX
			1000	1.057		0.488	100.0	0.189	16.4.0	0.581
2 7 10	21-12	10	10000	1.030	300-0	0.723	0.181	0.211	0.505	0.618
280	23-2	15	1000.	0.456	0.283	0.768	0.314	0.240	0.517	0.674
281	23-2	18	1600.	0.928	0.276	0.788	0.373	0.253	0.524	0.769
2.82.1	1-02	35	1000.	0.695	0.361	0*130	0.407	0.210	0.439	0.798
282.2	21-1	332	1000.	0.695	0.361	0*730	0*401	0.270	0.439	338.0
282.3	2-12	35	1000.	0.734	0.289	0.828	0.505	0.359	110.0	0.844
282.4	51-3	26	1000*	C.176	0. 302	561 *0	0.410	167.0	0.22	5 5 D * D
262.5	23-2	-16	1000.	0.891	0.260	0.825	0. 48.0	0.217	0.440	0.877
283	1-12	5	1000.	0.100	0.010	1 7 8 3	144.0	0.111.0	0.450	0.838
4.00	1-02	0.4	1600.	0.017	0.251	0.564	0.317	0.321	675.0	0.474
1 400	21-2	25	1505-	0.935	0.256	0.554	0.301	0.314	0.544	0.473
286.2	1-12	26	1505.	1.003	0.261	0.527	0.242	0.221	0.539	614.0
286.3	23-2	31	1505.	1.090	0.241	0.538	0.267	0.227	0.559	0.472
286.4	24-2	31	1505.	1.090	0.241	0.538	0.251	0.227	0.559	0.450
287	2-12	35	1505.	0.904	0*235	0.546	0.383	0*343	0.000	626.0
288.1	1-02 .	35	1495.	0.845	0.258	0*581	0.346	0.324	195*0	0*201
288.2	20-5	34	1495.	0.791	0.263	0.589	0.343	0.425	160.0	100.00
298.3	20-3	26	1495.	0.933	0.220	0.017	105-0	112.0	0.000	100.0
228.4	20-4	32	1495.	0.887	602*0	**0*0	114.0	0.084	160.0	100.00
268.5	2-12	50	1495.	0.887	0*203	*****	1	0. 344	110.0	0.000
289.1	20-1	35	.5061	0.820	0.102	0.548	01 6 30	C + 1 3	0.404	0.441
283.2	2-12	\$ 7	1200.	0,000	1.1.1	0.898 0.898	0.246	01010	0.673	0.429
289.2	1-42	15	1500	0.804	0.215	0.650	0.671	168.0	0.585	0.694
	1-12	25	1809.	0.805	0.215	0.650	0.671	0.391	0.585	0.734
24013	1-10	5.5	1505.	0.853	0.150	0.728	0.492	11.0	0.650	0.734
1.162	18-1	34	1510.	0.685	0.256	0.622	0.417	0.272	0.544	0.742
241.2	21-1	35	1510.	0.778	0.179	0.702	0.610	0.438	0.621	0.802
292	1-81	34	1495.	0.704	0.133	0.176	0.538	0.367	0.659	0.768
593	21-2	3.5	1005.	1.789	0.148	0.746	0.135	0.319	0.612	0.735
294.1	2-12	35	1005.	1.11	0.165	0.807	0.240	0. 281	0.034	0.141
2.462	2-62	16	1005.	1.958	0.160	0. 703	\$07*0	0.058	0*0*0	0.140
2.94.3	2 47	10	1000.	50%-1	0.145	0.865	0.426	144.0	0.655	0.891
1.020	03-2	i.	1000.	1.924	0.136	0.851	0.359	0.305	0.664	0.890
546.3	24-2		1000.	1.924	0.135	0.851	658.0	0.305	0.664	0.847
296	24-52	31	1000.	1.861	0, 345	\$56.0	0.612	0.380	0.705	0.929
1.162	8-3	26	1005.	1.630	0.082	1.003	0.772	0.421	0.718	1.086
297.2	21-2	35	1005.	1.596	0.073	1.023	0.874	0.629	0.727	1.086
298	21-2	35	1495.	1.877	0.112	0.034	0.361	0.433	0.034	185-0
299.1	20-3	26	1505.	1.981	0.103	700.0	0.515	0.310	140.0	0.034
2*662	21-2	22	*00	1021	05050	10.00	0.613	10 202	1112.0	0.638
5-662	2-42		*****	120.12	1 044	0.725	0.44	04.4.0	0.756	0.798
1.005	2-2	12	1510.	1.834	0.081	0.122	0.669	0.351	0.719	0.148
2*000	6-16	35	1510	1.760	0.073	0.745	0.744	0.527	0.727	0.798
4.005	21-3	26	1510.	1.834	0.081	0.722	0.669	0,351	611.0	0.798
301	8-3	26	1500.	1.826	0.029	0.838	0.882	105.0	0.771	0.877

LCCAL SUBCHANNEL CHE CONCITIONS (ENGLISH UNITS) ANC2 DATA

	CHF ROD NO.	CHF CHANNEL	PRESSURE (PSIA)	MASS VEL. (LBM/HR-F12) X 10-6	EQUILIERIUM QUALITY	9 8 E (81 40F12	0 1 C T U/HR-FT BEM-2	E D 21 X 10 BARNETT	с н ғ -6 6€	MEASURED CHF (BTU/HR-FT2) X 10-6 HFLUX
1 2 1 0 1	8-2	27	1505.	1.947	-0.051	1.001	1.252	0.802	0.851	0.978
	8-3	56	1505.	1.780	-0.011	0.917	1.044	0.487	0.811	0.978
No. No. <td>1-92</td> <td>36</td> <td>1005</td> <td>0.594</td> <td>0.527</td> <td>0.484</td> <td>0.042</td> <td>0.108</td> <td>0.273</td> <td>0.445</td>	1-92	36	1005	0.594	0.527	0.484	0.042	0.108	0.273	0.445
30-1 11 1000, 1	74-1	16	-565	0.555	164.0	0.567	0.243	0.106	0.268	0.509
11-2 19 1999. 0.4461 0.4473 0.4431	24-1	31	1006.	0.50%	0.465	0.644	0.451	0.180	0.335	0.572
X1-3 X6 1993. 0.4511 0.4431 0.4431 0.4130 0.4119 0.4111 0.4111	21-2	3.5	1495.	0.465	0.428	0.463	0.179	0.222	0.412	0.390
201-1 35 1905. 0.4411 0.4401	21-3	26	1495.	0.513	0.433	0.433	0.106	0.176	0.367	0. 390
1.1 1	1-02	35	1505.	0.443	0.430	0.467	661.0	0.225	0.288	0.473
21-1 35 1100. 0.4181 0.4422 0.4012 0.4012 0.4013	21-1	35	1505.	644.0	0.430	194.0	0.193	0.225	0.238	105.0
21-1 35 1010- 0.441 0.400 0.441 0.400 0.420 0.4	-1-12	36	1500.	0.393	0.422	205-0	0.274	247*0	0.288	055.0
20-1 15 1005. 0.4486 0.5114 0.6750 0.4146 0.5114 0.4759 0.4166 20-1 15 1005. 0.4486 0.4418 0.4514 0.4193 0.4269 0.4396 20-1 15 1005. 0.4418 0.4112 0.4491 0.4269 0.4193 0.459 0.4193 0.459 0.4193 0.459 0.4193 0.459 0.4193 0.459 0.4193 0.459 0.4193 0.459 0.4193 0.4193 0.4193 0.4113 0.4114 0.459 0.4113 0.4114 0.459 0.4113 0.4114 <td< td=""><td>21-1-</td><td>35</td><td>1510.</td><td>0.415</td><td>0.337</td><td>0.605</td><td>0.443</td><td>0.307</td><td>0.503</td><td>640*0</td></td<>	21-1-	35	1510.	0.415	0.337	0.605	0.443	0.307	0.503	640*0
21-1 15 1005. 0.4710 0.4711	20-1	36	1005.	0.386	0.514	0.626	0.717	0.144	0. 326	669.0
24-1 31 1005. 0.4473 0.4401 0.4504 0.4913 0.4913 0.4914 0.4913 0.4914	21-1	35	1005.	0.386	0.514	0.626	0.717	0.199	0.326	0.693
71-1 15 1505. 0.4412 0.40111 0.40111	24-1	- 31	1005.	644*0	0.459	0.666	0.544	0.193	0.381	0.658
20-1 15 1495. 0.4410 0.432 0.4431 0.4410 0.4413 24-1 11 1095. 0.4410 0.4413 0.4413 0.4413 0.4413 24-1 11 2005. 1.993 0.4113 0.4755 0.4913 0.413 24-2 2005. 1.996 0.4013 0.4913 0.4913 0.4913 24-2 2005. 1.996 0.4013 0.4913 0.4913 0.4913 24-2 2005. 1.996 0.4013 0.4913 0.4913 0.4913 24-2 2005. 1.996 0.4013 0.4913 0.4913 0.4913 24-1 27 2005. 1.4913 0.4013 0.4913 0.4913 24-1 27 2005. 1.4913 0.4013 0.4913 0.4913 24-1 27 2005. 1.4913 0.4013 0.4913 0.4913 24-1 27 2005. 1.4913 0.4013 0.4913 0.491	21-1	35	1505.	0.432	0.311	0.633	164*0	0.328	0.529	0.658
34-1 1 1445.	20-1	35	1495.	0.415	0.392	0.538	0.326	0.263	0.448	0.619
27-2 27 2005. 1.938 -0.011 0.4726 0.4942 0.5545 0.4942 0.5756 0.4719 0.7775 27-2 2005. 1.9901 0.4011 0.4726 0.4973 0.4973 0.4775 0.4973 0.4775 0.4875 <td< td=""><td>24+1</td><td>11</td><td>1495.</td><td>0.538</td><td>0.359</td><td>0.547</td><td>0.292</td><td>0.223</td><td>0.441</td><td>0.623</td></td<>	24+1	11	1495.	0.538	0.359	0.547	0.292	0.223	0.441	0.623
27: 2005. 1.441 0.017 0.465 1.442 0.472 0.474 0.473 0.474 0.473 8:-2 27 2005. 1.401 0.0119 0.465 1.442 0.473 0.473 0.473 8:-2 27 2005. 1.470 0.0173 0.465 0.473 0.4013 8:-1 27 2005. 1.470 0.0173 0.4613 0.4613 0.4613 8:-1 27 2005. 1.473 0.473 0.4613 0.4613 8:-1 27 2005. 1.471 0.473 0.4613 0.4613 8:-1 27 2005. 1.473 0.473 0.473 0.473 8:-1 27 2005. 1.441 0.473 0.473 0.473 8:-1 27 2005. 1.443 0.473 0.473 0.483 8:-1 27 2005. 1.441 0.473 0.493 0.483 8:-1 273	8-2	2.2	2005.	1.958	\$10.0-	0.726	246.0	0.596	0.813	0.775
6-2 27 2005. 0.079 -0.466 1.442 0.4766 0.4679 0.4674 8-1 27 2005. 1.976. -0.113 0.4766 0.4786 0.4913 0.4564 8-1 27 2005. 1.976. -0.113 0.4786 0.4913 0.4564 8-1 27 2000. 1.231 0.4014 0.4512 0.4913 0.4524 8-1 27 2000. 1.231 0.4014 0.4512 0.4913 0.4524 8-1 27 2000. 1.231 0.4014 0.4913 0.4913 0.4617 8-1 27 2005. 1.4387 -0.4175 0.4811 0.4714 0.4714 0.4723 0.4913 8-1 27 2005. 1.4381 -0.4175 0.4811 0.4714 0.4714 0.4714 8-1 27 2005. 1.4493 -0.4175 0.4911 0.4714 0.4913 0.4913 8-1 27 2005.	25-5	35	2005.	1+841	0.016	0.665	0.849	0.542	0.784	0.772
P-2 27 2005. 1.970 -0.113 1.0226 1.440 0.479 0.4913 0.427 R-1 27 2005. 1.970 -0.113 1.025 0.493 0.4973 0.4073 R-1 27 2005. 1.270 0.4074 0.4511 0.4571 0.4571 0.4571 0.4573 0.4073 R-1 27 2005. 1.1293 0.4074 0.4511 0.4511 0.4573 0.4573 0.4573 R-1 27 2005. 1.491 0.4171 0.4511 0.451 0.4513 0.4573 R-1 27 2005. 1.491 0.4111 1.471 0.472 0.493 0.493 R-1 27 2005. 1.491 0.4111 0.471 0.493 0.493 0.471 R-1 27 2005. 1.491 0.4112 0.491 0.411 0.472 0.493 0.493 0.493 R-1 27 10050 1.441 0.4	8-2	2.2	2005.	2.020	-0.079	0.866	1+142	0.726	0.879	0.854
8-1 27 2005 1.975 -0.113 1.602 1.382 0.4515	2-2	2.7	2005.	1.960	-0.113	0.926	1.240	0.190	616.0	0.927
8-1 27 2000. 1.884 0.4014 0.4578 0.4524 0.4786 0.4577 0.4614 0.4617 0.4674 0.4614 0.4617 0.4614 0.4617 0.4717 0.4717 0.4717 0.4717 0.4717 0.4612 0.4812	8-2	2.7	2005.	1.970	-0.153	1.002	1.362	0.869	0.953	1.003
18-1 14, 2000. 1.250 0.4096 0.4557 0.4311 0.7704 0.4657 0.4311 0.7704 0.4657 0.4311 0.7704 0.4657 0.4311 0.704 0.4573 0.4311 0.704 0.4573 0.4313 0.4573 0.4513 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4523 0.4313 0.4733 0.4533 0.4313 0.4523 0.4313 0.4523 0.4313 0.4533 0.4313 0.4533 0.4313 0.4323 0.4313 0.4533 0.4313 0.4323 0.4313	3-1	27	2000.	1.384	0.014	0.678	0.828	0*554	0.186	0.677
18-1 34 0.001 0.401 0.401 0.401 0.402 0.433 0.473 0.4	18-1	34	2000.	1.250	0.096	155.0	0.581	0.311	0.104	0.627
H-1 Z7 Z000. L-387	13-1	34	2000.	1+231	0.007	0. 604	0.652	0*333	0.733	269*0
8-2 27 2005. 1.4498 -0.172 0.4881 1.0.773 0.4925 0.4925 0.4875 0.4925 0.4875 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975 0.4975	8-1	27	2000.	1.387	-0.024	0.735	166.0	0.613	0.824	0.147
R-1 27 2405. 1.453 4.075 0.4112 0.477 0.477 0.475 0.473 0.475 0.473 R-1 77 1995. 1.453 0.125 0.473 0.473 0.475 0.483 0.493 <td>8-2</td> <td>27</td> <td>2005.</td> <td>1.498</td> <td>-0.122</td> <td>0.881</td> <td>1.210</td> <td>0.173</td> <td>776.0</td> <td>0.823</td>	8-2	27	2005.	1.498	-0.122	0.881	1.210	0.173	776.0	0.823
R-1 77 1995. 1.451 -0.125 0.475 1.217 0.4725 0.4830 0.4830 0.4813 18-1 77 2005. 1.431 0.0130 0.4716 0.4933 0.4830 0.4813 7 2005. 1.430 -0.039 0.4716 0.4733 0.4830 0.4813 7 2005. 1.430 -0.112 0.4716 0.4711 0.4732 0.4830 0.4813 7 2005. 1.430 -0.112 0.4716 0.4711 0.4728 0.4912 0.4912 0.4913 7 2000. 1.4410 0.4112 0.4716 0.4516 0.4122 0.4913 7 2000. 1.4410 0.4112 0.4710 0.4516 0.4912 0.4913 8-2 2000. 1.4712 0.4710 0.4710 0.4912 0.4913 8-2 2000. 1.4934 1.1095 1.4910 0.4912 0.4912 8-2 2000. 1.4934	1-1	27	2005.	1.459	-0.075	0.812	1.076	0.044	0.875	0.823
16-1 34 1995 . 1.284 -0.030 0.773 0.9433 0.9433 0.9433 0.9413 $8-1$ 37 2005 . 1.2845 -0.0162 0.9723 1.3913 0.9912 0.9912 $8-1$ 37 2005 . 1.4413 -0.0162 0.9723 0.9912 0.9912 34 2000 . 1.4413 -0.1122 0.9726 0.9912 0.9912 34 2000 . 1.4413 -0.1256 0.4734 0.912 0.9972 34 2000 . 1.311 0.9274 0.912 0.9972 $8-2$ 27 2000 . 1.934 -0.126 0.1714 1.1042 0.9272 0.9972 $8-2$ 27 2000 . 1.934 0.1114 1.1093 1.134 0.912 0.9273 0.9972 $8-2$ 27 2000 . 1.934 1.1093 1.1094 1.1134 0.9121 0.9282	8-1	10	1995.	159*1	-0.125	0.382	1.219	111.0	0.925	0.878
R-1 77 2005. 1.430 -0.162 0.922 1.305 0.9450 0.9462 0.9463	18-1	34	1995.	1.284	-0.030	0.743	0.905	0.433	0.830	0.813
18-1 34 2005 1.285 -0.358 0.776 0.577 0.6858 0.976 0.976 $2+1$ 36 2000 1.4404 0.112 0.826 1.311 0.743 0.912 0.975 $2+1$ 34 2000 1.4410 0.112 0.826 1.311 0.743 0.912 0.975 $2+1$ 34 2000 1.274 0.112 0.826 0.748 0.912 0.997 $8+2$ 277 2000 1.1224 -0.1214 1.095 1.457 0.912 0.923 $8+2$ 277 2000 1.378 -0.1214 1.095 1.134 0.114 1.134 $8+2$ 277 2095 2.433 -0.1612 0.951 0.923 0.9247 $8+2$ 277 2005 2.433 -0.1214 1.134 0.711 0.921 0.923 $8+2$ 277 2005 <td>9-1</td> <td>22</td> <td>2005.</td> <td>1.430</td> <td>-0.162</td> <td>0.922</td> <td>1.309</td> <td>0.830</td> <td>0.962</td> <td>0.980</td>	9-1	22	2005.	1.430	-0.162	0.922	1.309	0.830	0.962	0.980
75-1 36 2005. 1.4404 0.112 0.4956 1.171 0.774 0.9712 0.997 3-1 27 2000. 1.4410 0.5155 0.772 1.5171 0.4956 0.9952 0.9957 3+1 34 2000. 1.2754 -0.112 0.8556 0.771 0.912 0.9923 0.9925 3+2 2000. 1.378 -0.112 0.8556 0.771 0.912 0.923 0.912 27 2000. 1.923 -0.112 0.854 1.169 0.778 0.912 0.923 8-2 27 2000. 1.923 -0.111 0.775 0.932 0.912 1.096 8-2 27 2005. 2.492 -0.111 0.775 0.9817 1.014 1.134 8-2 27 2005. 2.492 -0.111 0.710 0.9810 0.912 0.923 8-2 27 2005. 2.492 -0.1111 0.710 0.911	1-81	34	2005.	1.285	-0.358	0.176	0.870	0.458	0.858	876*0
n-1 77 2000 1.4410 2.162 0.9701 0.972 0.9702 0.9702 $18-1$ 34 20000 1.254 -0.356 0.776 0.975 0.9702 0.9702 $28-2$ 27 2000 1.378 -0.1167 1.053 1.460 0.932 0.992 0.992 $8-2$ 27 2000 1.923 -0.1167 1.053 1.460 0.932 0.997 1.096 $8-2$ 27 2009 1.923 -0.214 1.090 1.537 0.997 1.014 $8-2$ 27 2009 2.442 -0.2101 0.776 0.912 1.096 $8-2$ 27 2009 2.442 -0.2111 0.780 0.912 1.014 $8-2$ 27 2009 2.442 -0.2111 0.780 0.911 1.1234 $8-2$ 2005 2.442 -0.1111 0.710 <	25-1	36	2005+	1.4.94	0.112	0.826	1.1.1.1	6.41.0	216.0	227.0
$ \begin{array}{cccccccccccccccccccccccccccccccccccc$	9-1	27	2000.	1.410	2.162	0* 420	1.511	0.829	294.0	266.0
25-1 56 2000 1.378 -0.1112 0.4874 1.1694 0.716 0.981 1.034 $8-2$ 27 2000 1.923 -0.187 1.053 1.464 0.932 0.981 1.034 $8-2$ 27 2095 2.442 -0.214 1.053 1.014 1.134 $8-2$ 27 2095 2.442 -0.2101 0.770 0.981 0.810 $8-2$ 27 2095 2.443 -0.111 0.710 0.810 0.992 1.134 $8-2$ 27 2005 2.433 -0.111 0.713 0.710 0.810 0.992 1.1234 $8-2$ 277 2005 2.431 -0.1231 1.1625 1.023 1.1234 $8-1$ 277 2005 2.431 -0.1231 1.427 0.911 0.273 $8-1$ 277 2005 2.345 1.073 $1.$	1-11	34	2000.	467*1	-0*02Q	0.113	* * * *	00110	0.00	0.74.0
8-2 27 2000. 1.923 -0.107 1.003 1.934 0.981 1.014 1.134 8-2 27 2095. 1.884 -0.214 1.099 1.537 0.981 1.014 1.134 8-2 27 2095. 2.443 -0.011 0.710 0.587 0.5810 0.610 0.683 8-2 27 2095. 2.443 -0.011 0.711 0.710 0.864 0.810 0.683 8-2 27 2095. 2.433 -0.011 0.713 0.810 0.911 0.713 0.810 0.784 0.810 0.784 0.713 0.713 0.713 0.713 0.713 0.713 0.713 0.713 0.713 0.723 1.1234 0.713 0.723 0.723 1.1234 0.723 0.713 0.723 0.723 0.723 0.723 0.723 0.723 0.723 0.723 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.725 0.7	25-1	36	2000.	1.378	-0.112	0.854	1.109	2410	216.0	744.0
R-2 27 1935. 1.884 -0.214 1.090 1.937 0.981 1.014 1.154 R-2 27 2005. 2.443 -0.010 0.775 0.987 0.810 0.823 R-2 27 2005. 2.443 -0.010 0.775 0.987 0.810 0.823 R-2 27 2005. 2.433 -0.111 0.981 1.478 0.911 0.993 R-2 27 2005. 2.433 -0.1111 0.981 1.412 0.911 0.993 R-2 27 2005. 2.433 -0.1111 0.981 1.412 0.911 0.993 R-1 27 2005. 2.431 -0.1111 0.911 1.278 0.911 1.234 R-1 27 2005. 2.451 0.0171 0.462 0.967 0.963 1.234 R-1 34 2000. 0.872 0.035 0.726 0.759 0.759 18-1 34 2000. 0.873 0.713 0.726 0.759 0.653 0.653	2-6	- 27	2000.	1.925	-0-54	1.033	1.400	0* 435	194.0	040*T
8-2 27 2005. 2.447 -0.010 0.175 0.787 0.510 0.964 8-2 27 2005. 2.492 -0.0104 0.869 0.710 0.864 0.997 8-2 27 2005. 2.492 -0.111 0.711 0.864 0.911 0.997 8-2 27 2005. 2.433 -0.1111 0.712 0.911 0.912 8-2 27 2005. 2.433 -0.1111 0.712 0.911 0.912 8-1 27 2005. 2.451 -0.152 1.131 1.412 0.913 0.913 8-2 27 2005. 2.351 -0.153 1.131 1.497 0.913 1.234 8-2 27 2005. 2.351 -0.0710 0.4626 0.402 0.933 1.234 18-1 34 2700. 0.872 0.713 0.773 0.773 0.551 18-1 34 2000. 0.887 0.0163 0.773 0.759 0.643 18-1 34 2003.	8-2	27	1995.	1.834	-0.214	1.090	1.56.1	186.0	1.014	1.134
8-2 27 2005. 2.492 -0.1064 0.867 1.1134 0.710 0.664 0.997 8-2 27 2005. 2.433 -0.111 0.9181 1.234 0.710 0.693 8-2 27 2005. 2.433 -0.111 0.4911 1.442 0.911 0.911 8-2 27 2005. 2.451 -0.152 1.1131 1.442 0.913 0.152 8-1 27 2005. 2.351 -0.152 1.1131 1.447 0.913 0.1234 8-2 27 2005. 2.351 -0.123 1.1218 1.625 1.026 1.234 18-1 34 2005. 2.361 -0.233 0.513 0.713 0.375 0.759 18-1 34 2000. 0.875 0.0313 0.6526 0.755 0.757 0.598 18-1 34 2000. 0.8764 -0.033 0.7657 0.779 0.643 18-1 34 2000. 0.875 0.795 0.767 0.6598 18-1 34 2000. 0.8764 0.795 0.767 0.6598 18-1 34 2000. 0.8764 0.795	2-2	27	2095.	644*2	-0.013	0.725	156.0	0.387	0.810	0.863
B-2 27 2005. 2.433 -0.111 0.481 1.278 0.614 0.911 0.991 R-2 27 2005. 2.430 -0.152 1.078 1.412 0.967 0.993 1.106 R-2 27 2005. 2.350 -0.152 1.078 1.412 0.9657 0.993 1.234 R-1 27 2005. 2.351 -0.183 1.218 1.412 0.9657 1.023 R-1 27 2005. 2.351 -0.223 1.218 1.423 1.234 18-1 34 2000. 0.872 0.071 0.652 0.759 0.759 18-1 34 2000. 0.872 0.033 0.703 0.795 0.759 18-1 34 2000. 0.876 0.033 0.793 0.759 0.554 18-1 34 2000. 0.876 0.793 0.795 0.759 0.759 18-3 26 0.703 0.793 0.793 0.759 0.759 18-1 34 2000. 0.9804 -0.003 0.793 0.756 18-3 27 2000. 0.9804 -0.0163 0.793 0.756 <tr< td=""><td>5-8</td><td>22</td><td>2005.</td><td>264.2</td><td>-0.064</td><td>0.863</td><td>1+134</td><td>0.110</td><td>0.804</td><td>969.0</td></tr<>	5-8	22	2005.	264.2	-0.064	0.863	1+134	0.110	0.804	969.0
9-2 27 2000. 2.430 -0.152 1.078 1.412 0.498 0.492 1.100 8-1 27 2005. 2.350 -0.1183 1.1131 1.497 0.983 1.234 8-2 27 2005. 2.361 -0.223 1.131 1.497 0.983 1.234 18-1 34 2005. 2.361 -0.223 1.518 1.625 1.079 0.795 18-1 34 2000. 0.872 0.071 0.655 0.713 0.759 0.551 18-1 34 2000. 0.876 0.033 0.667 0.713 0.759 0.558 18-1 34 2000. 0.876 0.033 0.667 0.795 0.558 18-1 34 2000. 0.864 -0.003 0.703 0.795 0.556 18-1 34 2000. 1.025 -0.163 0.795 0.535 0.563 0.758 8-3 26 2000. 0.980 -0.033 0.793 0.795 0.758 8-3 2000. 0.980 -0.033 0.793 0.795 0.758 8-3 2000. 0.902 0.992 0.793 0.	3-2	- 22	2005.	2+433	-0.111	185.0	1.278	0.814	116.0	164.0
R-1 27 2005 2.350 -0.183 1.131 1.497 0.967 0.483 1.234 8-2 27 2005 2.361 -0.223 1.218 1.625 1.023 1.234 18-1 34 2000 0.872 0.071 0.657 0.779 0.779 0.596 18-1 34 2000 0.875 0.071 0.657 0.773 0.779 0.598 18-1 34 2000 0.876 0.033 0.673 0.773 0.799 0.598 18-1 34 2000 0.864 -0.003 0.6703 0.773 0.795 0.598 18-1 34 2000 1.025 -0.163 0.703 0.795 0.595 8-3 26 0.703 0.793 0.793 0.795 0.558 8-3 2600 1.025 -0.163 0.895 1.066 0.758 8-2 2000 0.902 -9.2766 0.897 1.463 0.775 0.758	2-2	27	2000.	2.430	-0*152	1.078	1.412	0.408	266.0	1.100
8-2 27 2005, 2.361 -0.223 1.218 1.625 1.026 1.023 18-1 34 2000, 0.872 0.071 0.6526 0.656 0.729 0.551 18-1 34 2000, 0.875 0.0333 0.6526 0.6734 0.729 0.558 18-1 34 2000, 0.876 0.0333 0.6733 0.713 0.729 0.558 18-1 34 2000, 0.864 -0.0033 0.7163 0.779 0.6643 0.643 18-1 34 2000, 1.025 -0.163 0.785 0.795 0.643 0.643 8-3 26 2000, 1.025 -0.163 0.785 1.188 0.535 0.643 8-3 27 2000, 0.902 -9.266 0.880 1.465 0.725	3-1	27	2005 .	2+350	-0.183	1.131	169-1	196.0	0.483	1.234
18-1 34 2000. 0.872 0.071 0.626 0.566 0.713 0.779 0.551 18-1 34 2000. 0.876 0.033 0.667 0.713 0.375 0.757 0.598 18-1 34 2000. 0.864 -0.003 0.603 0.703 0.713 0.375 0.757 0.598 18-1 34 2000. 0.864 -0.003 0.703 0.773 0.803 0.643 18-1 34 2000. 1.025 -0.163 0.703 0.795 0.643 8-3 26 2000. 1.025 -0.163 0.895 1.188 0.535 0.963 0.725 8-3 27 2000. 0.992 -0.163 0.880 1.463 0.872 1.066 0.758	8-2	27	2005.	2.361	-0.223	1.218	1.625	1.056	1.023	1.234
IB-1 34 2000. 0.876 0.033 0.667 0.713 0.375 0.767 0.598 IR-1 34 2000. 0.864 -0.003 0.703 0.795 0.643 0.643 IR-1 34 2000. 0.864 -0.003 0.703 0.795 0.643 0.643 R-3 26 2000. 1.025 -0.163 0.895 1.188 0.535 0.963 0.643 R-3 27 2000. 1.025 -0.163 0.880 1.463 0.372 0.728 A-2 27 2000. 0.992 -0.226 0.880 1.463 0.758	1-81	34	2000.	0.872	0.071	0.626	0.625	0.346	0+129	144.0
IR-1 34 2000. 0.864 -0.003 0.703 0.402 0.803 0.643 R-3 26 2000. 1.025 -0.163 0.852 1.188 0.355 0.463 0.725 R-3 26 2000. 1.025 -0.163 0.8852 1.188 0.355 0.463 0.725 R-2 27 2000. 0.992 -0.266 0.880 1.463 0.872 1.066 0.758	18-1	34	2000.	0.876	0.033	0.667	0-713	0.375	0.757	0.598
8-3 26 2000, 1.025 -0.163 0.852 1.188 0.335 0.953 0.725 8-2 27 2000, 0.902 -0.266 0.880 1.463 0.872 1.066 0.758	18-1	34	2000.	0.864	~0.003	0.703	0. 795	0*402	0.803	0.643
A-2 27 2000. 0,902 -0.266 0.880 1.463 0.872 1.066 0.758	m-8	26	2000.	1.025	-0.163	0.852	1.188	0.535	0. 463	621.0
	A-2	27	2000.	206*0	-3*266	0.880	1.463	0.872	1+066	951.0

LOCAL SUBCHANNEL CHE CONDITIONS TENGLISH UNITST ANC 2 DATA

	10-6																			
MFASURED CH	181U/HR-F121 X	HFLUX	日本日本日本 方子子	0.584	0.601	0.626	0.657	0.667	165*0	0.570	0.628	0.683	0.729	0.179	0.830	0.830	0.838	0.838	0.564	615.0
CHF	9-	99	三丁三三百	0.740	0.871	0.928	216.0	6.973	0.664	0.764	0.815	0.852	0.901	196.0	1.052	466-0	1.078	610-1	0.706	0*962
5 0	21 × 10	BAKNETT		0.486	0.442	0.486	0.516	0.515	0.305	0.509	565.0	0.661	0.496	0.533	0.910	0.580.	116.0	0.603	0.454	264*0
DICT	U/HR-FT	86W-2		0.661	626*0	1.064	1.164	1.156	0.474	2.715	0.888	1.013	1.078	1.176	1.488	1,301	1.589	1.361	0.595	0.677
PRE	181	1011		0.649	0.754	\$62.0	\$18.0	0.814	0. 580	0.669	0.719	0.777	0.824	0.869	0.434	+16.0	166.0	6.933	0.593	0.655
	EQUIL TBATUM	QUALITY		0.050	-0.071	-0.128	-0.173	-0.173	0.136	0.036	-0.015	-0.062 -	-0.101	-0.141	-0.252	-0.144	-0.278	-0.219	. 0* 394	0.053
MASS VEL .	(LBN/HR-F12)	x 10-6		0.687	0.768	0.813	0.792	0.795	0.646	0.683	1.165	1.197	1.273	1.231	1.113	1.213	1.254	1.206	0*893	0.685
	PRESSURE	(PS1A)		2005.	2000.	2000.	2000.	2005.	2005.	2005.	2000.	2000.	2000.	1995.	2000.	2000.	2000.	2000.	2010.	2005.
	CHF -	CHANNEL		36	26	26	20	26	34	3.5	27	27	26	26	27	26	- 22	26	36	3.6
	CHE	RCU 1.0.		25-1	8-3	8-3	8-3	8-3	25-1	29-1	8-1	8-1	8-3	8-3	8-2	8-8	2-8	8-3	25-1	25-1
		RUN NO.	*******	193	190	145	261	203	204	226	235	240	247	252	257.1	257.2	259.1	259.2	321	334

LOCAL SUBCHANNEL CHF CONDITIONS (ENGLISH UNITS) ANC 3 DATA

. .

RUN NO.	CHF RED NC.	CHF CHANNEL	PRESSURE (PSIA)	MASS VEL. (LBM/HR-FT2) X 10-6	EQUILIBRIUM QUALITY	PRE (BT LOFT2	0 1 C 1 UZHR-F1 86W-2	E D 121 X 10 BARNETT	CHF -6 GE	MEASURED CHF (BTU/HR-FT2) X 10-6 HFLUX
		31	2000	2 0 7 8	0.116	0.419	0.531	0.235	0.684	0.495
25	23-2	24	2000.	2.010	0.049	0.540	0.665	0.300	0.731	0.610
28	23-2	51	2005.	2+013	0.009	0.637	0.610	0.360	0.438	0.636
38	21-3	26	2005.	1.294	0.104	0.569	0.570	0.427	0.696	0.430
42.1	21-1	35	2000.	1.207	0.104	0.633	0.525	0.327	0.696	0.613
42+2	23-1	31	2000.	1.210	S+110	0.722	0.323	0.201	0.004	0.703
45.1	8-2	26	2015.	1.1/3	-0.019	0.719	0.851	0.413	0.819	0.782
45.2	7-2	26	2015.	1,173	-0.019	0.119	0.851	0.413	0-814	0.778
45.3	21-1	35	2015.	1.075	0.030	0.658	0.758	0.528	0.770	0.782
48	8-2	26	2000.	1.189	-0.089	0.807	1.034	0.480	0.889	0.878
51	8-2	26	2000.	1.172	-0.107	0.824	1.076	0.496	0.907	0.903
57	8-2	26	2000.	1.983	-0.030	0.764	0.950	0.430	0.830	0.853
6.2	8-2	26	2000.	2.014	-0.074	0.857	1.078	0.487	0.874	1.046
67	, 2-2	30	2000.	1.955	-0.146	0.988	1.337	1.853	0.946	1.139
70	12-2	30	2000.	1.925	-0.182	1.046	1.354	1.919	0.982	1.165
80	12-1	30	2005.	1.384	0.062	0.599	0.696	0.478	0.738	0.586
99	21-2	35	2000.	2.412	0.073	0.500	0.691	0.400	0.727	0.563
100-1	21-2	35	2000.	2.348	0.041	0.592	0.790	0.474	0.759	0.690
100.2	21-3	26	2000.	2.471	0.048	0.567	0.740	0.314	0.752	0.690
107	14-2	31	2000.	2.503	-0.047	0.826	1.033	0.457	0.820	1.038

and the second distance of
LOCAL SUBCHANNEL CHE CONDITIONS (ENGLISH UNITS) ANC4 DATA MEASURED CHF (BIU/HR-FI2) X 10-6 HFLUX * 1 0.5550.0 0.645 35 P. R. E. D. I. C. T. E. D. C. (STU/HR-FT2) X 10-6 LOFI2 SCM-2 BARNEIT 0.346 0.091 0.404 0.229 0.088 0+272 361-0 1.186 0+239 \$12* 1.330 442=0 0.307 0.472 0.144 0.220 0.266 465*0 0.352 0.494 0.372 0.424 .195 162.0 0.460 0.306 0.174 0.232 0.352 0.244 0.431 0.242 0.674 0.299 556*0 0.358 0.742 0.150 0.284 0.155 0.373 789.0 0.162 0.902 0.237 0.523 0.596 E10.1 0.533 0.261 664*0 0.363 0.4468 9.339 0.855 0.674 0.313 0.641 44.0 0.683 0.755 0.755 0.582 0.582 0.573 0.573 0.570 0.577 0.644 0.7755 0.717 0.755 0.537 0.537 EQUILIBRIUM 0.547 0.547 0.547 0.547 0.548 0.412 0.412 0.412 0.135 0.297 0.297 0.304 0.117 0.117 0.165 0.165 0.117 0.165 0.174 0.356 0.233 0.335 0.383 0.245 0.326 0.186 0.133 0.155 0.082 0.313 MASS VEL. (LBM/HH-FT2) E X 10-6 0.497 0.497 0.497 0.495 0.495 0.459 0.459 0.470 0.470 0.477 0.892 0.895 0.894 0.862 0.862 0.863 1.263 1.823 1.823 1.953 1.953 1.953 0.448 426*1 0*6*0 0.898 0.880 0.474 656.0 .928 0.455 946 *0 148*(3*963 PRESSURE (PSIA) 1005: 10005: 10005: 10005: 10005: 151 CHANNEL 16.16 RED NC. RUN NO.

APPENDIX C

STEADY STATE CHF CORRELATIONS

Predicted critical heat fluxes with the following steady state correlations were compared with the measured critical heat fluxes of the LOFT data.

1.	B&W-2	A . A (D 2000)
	(a -	$- b D_{e} [A_{1} (A_{2} G)]^{A_{3} + A_{4}(P-2000)} -A_{9}GX h_{fg}]$
	q =	$A_5(A_6 G)^{A_7} + A_8(P-2000)$
where	e q	= Critical Heat Flux (Btu/hr-ft ²)
	G	= Mass Velocity (Lbm/hr-ft ²)
	De	= Hydraulic Dia. (in.)
	Р	= Pressure (psia)
	hfg	= Latent Heat of Vaporization (Btu/1bm)
	Х	= Equilibrium Quality
and	а	= 1.15509
	b	= 0.40703
	A ₁	$= 0.37020 \times 10^8$
	A2	$= 0.59137 \times 10^{-6}$
	A3	= 0.8304
	A4	$= 0.68479 \times 10^{-3}$
	A ₅	= 12.71
	A ₆	$= 0.30545 \times 10^{-5}$
	A ₇	= 0.71186
	A8	$= 0.20729 \times 10^{-3}$
	A ₉	= 0.15208
2.	BARNETT	

$$q = \frac{\left[A \frac{h_{fg}}{649} - B h_{fg}X\right] \times 10^6}{C}$$

where

$$A = 69.4 D_{h}^{0.751} (Gx10^{-6}) (1.0 - 0.672 EXP [-6.09 (Gx10^{-6}) D_{W}])$$

$$B = 0.25 D_{h} (Gx10^{-6})$$

$$C = 165.9 D_{W}^{1.246} (Gx10^{-6})^{0.329}$$

and

q = Critical Heat Flux ($Btu/hr-ft^2$)

D_h = Heated Equivalent Dia. (in.)

 D_W = Wetted Equivalent Dia. (in.)

X = Equilibrium Quality

G = Mass Velocity (Lbm/hr-ft²)

hfg= Latent Heat of Vaporization (Btu/1bm)

3. GE

q =
$$10^{6} (0.8 - X)$$

FOR G $\ge 0.5 \times 10^{6}$
q = $10^{6} (0.84 - X)$
FOR G < 0.5×10^{6}

where

q = Critical Heat Flux (Btu/hr-ft²) X = Equilibrium Quality G = Mass Velocity (Lbm/hr-ft²) APPENDIX D

EXPERIMENTAL FACILITY DESCRIPTION

EXPERIMENTAL FACILITY DESCRIPTION

The LOFT experiments were conducted in the Medium Pressure Heat Transfer Flow Loop of the Chemical Engineering Research Laboratories at Columbia University. The experimental facility consists of: the heat transfer loop, the associated process instrumentation and control system, the dc power supply, the Computer Controlled Data Acquisition System (CCDAS), and the test section.

Heat Transfer Loop

The heat transfer loop was constructed of 300 Series stainless steel with the main piping of diameters 3- and 4-in. nominal. The loop is shown schematically in Figure D-1, and the principal loop operational limits are given in Table D-I. Table D-II gives the loop instrumentation for experimental parameters.

The major components of the loop are a 100 hp (75 kW) Wilson-Snyder centrifugal pump, the test section housing, mixing tee, heat exchangers, makeup system, and the purification system.

Primary Loop. The 100 hp (75 kW) Wilson-Snyder centrifugal pump provides circulation of the coolant around the closed primary loop. The total flow leaving the main circulating pump splits with part going through the test section and the remainder through the heat exchangers. The flow through the test section can be varied by means of a flowcontrol valve which is operated manually from the loop control panel area. A second flow-control valve in the heat exchanger line provides the test section inlet temperature control. The heat exchanger secondary flow, which is controlled manually from the loop control area,



Figure D-1 Schematic of Columbia University test loop.

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COLUMBIA MEDIUM PRESSURE HEAT TRANSFER LOOP OPERATIONAL LIMITS

16.5 MPa
0.035 m ³ /s
180 m of water
5.09 m
3.81 m
165 mm
23.2 m ²
0.038 m ³ /s
289 K

TABLE D-II

INSTRUMENTATION FOR THE COLUMBIA MEDIUM PRESSURE HEAT TRANSFER LOOP

Flow to test section [for flow above $0.0025 \text{ m}^3/\text{s}$]

- BIF Venturi, Serial No.
 6580-1, calibrated in BIF
 Hydraulic Laboratory.
 Differential pressure visually read on a 1524-mm
 Meriam mercury manometer.
- (2) Potter Aeronautical Corporation turbine flowmeter, 64 mm, Serial No. 2.5-1, read-out recorded on data acquisition system described below.

Heise gauge, 0 to 18 MPa with 14 kPa divisions marked, calibrated with a dead weight tester, Serial No. CMM-2395, read visually.

- Calibrated iron-constantan thermocouple displayed on Foxboro Dynalog Recorder Controller, Serial No. 673818, read visually.
- (2) Calibrated platinum resistance thermometer and bridge and/or calibrated iron-constantan thermocouple, recorded on data acquisition system.

Test section outlet pressure

Test section inlet temperature

TABLE D-II (Cont'd)

est section pressure drop	Visually read on three
	1524-mm Meriam man-
	ometers, using mercury and
	2.95 specific gravity nam-
	ometer fluid. Also measured
	using Barton differential
	pressure cells with elec-
	trical output recorded by
	data acquisition system.
est section power	Amperage from each generator
	measured using calibrated
	manganin shunts manufactured
	by the Westinghouse Elec-
	tric Corporation. Amper-
	age recorded with data
	acquisition system.
	Voltage measured across
	the main test section
	power terminals and across
	voltage taps built into
	heater rods. Voltage
	recorded with the data

Subchannel coolant temperature and test section outlet temperature

CHF detection

Iron-constantan thermocouples inside the heater rods, continuous readout on five 8-channel Offner Dynagraph recorders manufactured by the Beckman Company, and an 80-channel

acquisition system.

thermocouples, output

system.

Calibrated iron-constantan

recorded by data acquisition

TABLE D-II (Cont'd)

bar graph display manufactured by Metra Instruments, Inc. This instrument has an adjustable trigger mechanism which automatically shuts down the power to the test section in the event of a temperature excursion. provides additional control capability. The test section flow is measured by a venturi and a turbine flowmeter prior to its entry into the test section housing. Within the test housing, the coolant removes heat from the heated test section and exits from the top of the test housing to unite in the mixing tee with the flow from the heat exchangers. The mixing tee provides a stable inlet temperature at the pump inlet and hence at the test section inlet. The heat exchanger branch of the primary loop consists of three heat exchangers and an air-actuated control valve. The heat exchangers have a total heat transfer area of 23.2 m^2 with 15.4 m² in the main heat exchanger.

The loop is constructed in such a way that the heat exchangers can be operated singly or in any combination, thus providing a wide range of achievable subcooling even for low mass flow rates. The heat exchangers are tube-in-shell type with the primary loop water on the tube side. The cooling water for the heat exchangers is obtained from onsite wells. The secondary system is a once-through open loop.

Makeup, Feedwater, and Pressure Control System. The makeup and feedwater system is used to fill the loop initially and to provide makeup water during operation to maintain loop pressure. Whenever the loop pressure decreases, a Bristol Model 1G500 FF-14 pressure recorder controller activates an air driven makeup pump (piston pump) to restore the pressure to the set value. Whenever the reference pressure exceeds the set control value, the over pressure is relieved by a "Mighty Mite" relief valve in which the reference pressure acts on one side of a diaphragm and the loop water pressure on the other side. The makeup system uses deionized water which is stored in a heated [370 K]

intermediate tank. The intermediate storage tank is kept filled automatically by city water fed through the main deionizing system.

<u>Purification System</u>. The purification system is used to control the water chemistry in the primary loop. Part of the heat exchanger outlet flow is split into two streams. One stream is taken through the Graham Heliflow heat exchangers (Hx-2 and Hx-3 in Figure D-1) and is used to cool the pump seal, finally merging with the pump inlet flow. The second stream is taken through two Parker Dual Coil heat exchangers (Hx-4 and Hx-5) and again split into two streams. One stream passes through the deionizers and joins the pump suction stream. The second stream is used to cool one of the test section chamber seals and is returned to the pump suction stream.

<u>Well Water System</u>. Water from onsite wells is used as the heat sink in the loop heat exchangers and in the Parker Dual Coil heat exchangers (Hx-4 and Hx-5) and Graham Heliflow heat exchangers (Hx-2 and Hx-3) described in the purification system. The water from the wells is pumped by three centrifugal pumps, operated in parallel, to the inlet of the secondary side of the heat exchangers. The well water coming out of all the heat exchangers is discharged into the sewer.

DC Power System

The electrical power for heating of the test sections is obtained from the dc power system. The complete dc power system consists of four dc generators and the motors which drive them, motor-generator protective systems, control panels in the experimental areas for remote operation, instrumentation, and the interlocking system which controls

the equipment. Power for the dc power system is supplied by two highvoltage feeders from the nearby power substation.

The principal components of the dc power system are the motorgenerator sets. The two generators in the building basement have a single motor and two exciters, one supplying the fields of the two generators and the other the motor. The other two generators in the driveway area are normally operated in parallel and have independent motors. The exciter output is controlled by a semiconductor circuit, the output of which is determined by a 25-turn potentiometer in the test area, thereby allowing convenient adjustment of power to the test section. There are two potentiometers, each controlling the output of one pair of generators.

Figure D-2 shows a schematic of the dc power system for the medium and high pressure heat transfer flow loops at the laboratory. The power system specifications are summarized in Table D-III.

Computer Controlled Data Acquisition System

<u>Hardware</u>. The "Computer Controlled Data Acquisition System" (CCDAS) comprises:

- Hewlett-Packard, HP 2100 Computer 32K core, with two channel direct memory access, time base generator, and floating point hardware.
- (2) A digital magnetic tape unit, 9 track, 800 bpi, 37.5 in. per second.
- (3) HP 2401C digital voltmeter.



Figure D-2 DC power system.

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TABLE D-III

DC POWER SYSTEM SPECIFICATIONS

	Driveway Sets	Basement Sets	
	DC Continuously Variable		
Output:	-30 to 240 volts	-30 to 185 volts	
Current:			
Continuous rating	6000 amps each	9000 amps each	
Short time rating	8000 amps each	10,000 amps each	
Type:	Separately	excited	
Generator Speed:	26.9 rad/s (514 rpm)		
Ripple:	2% at 2100 Hz (2100 cps)		

- (4) HP 2322 subsystem, crossbar scanner and controller (200 channel, 3-wire input).
- (5) ASR 33 teleprinter.
- (6) HP 2762A Terminet Printer, 30 characters per second, 118 columns.
- (7) HP 2895B papertape punch unit.
- (8) HP 2748B papertape reader.
- (9) An HP 2313B subsystem with a programmable pacer, dual channel digital-to-analog converter and multiplexers for 112 channel differential input for high speed data acquisition (8000 samples per second for sequential channel scanning, 45 kHz single channel sampling). The computer is equipped with the necessary interfaces for the operation of the above subsystems.

The computer operates under a Magnetic Tape System (MTS). Experimental data are recorded on magnetic tape and are loaded into the core each time they are to be used.

Software. The fast data acquisition system (FDAS) software consists of a main program, LOFT3, which controls the use of a number of data reduction and acquisition subroutines. LOFT3 is in on-line contact with the test operator through the ASR 33, cathode ray tube (CRT), or the Terminet Printer and, according to the options chosen, will initiate single sequential scan, multiple-sequential scanning, scanning of a random block of channels, and either pre or posttest reduction of certain variables. The data reduction option picks the specified scan from the magnetic tape; reduces the data to engineering units (with appropriate

zero corrections); makes comparison of flow from turbine flowmeter and venturi meter readings; compares temperatures from thermocouple and resistance temperature detector (RTD) (any major deviations are flagged on the output); and calculated heat balance, temperatures, pressure drops, and test section resistances. The output is returned on the teleprinter or terminet. The data acquisition and control subroutines are:

- HSCAN: Performs single or multiple sampling of single channel or block or sequential channels of the multiplexer system, writes scanned data on tape, and/or prints raw data using the following HP system routines:
- DATIM: HP time of day routine
- SETUP: Sets up computer internal clock
- STCLK: Initiates the interval timer of the computer
- SCODE: An interpretation code for translating sequential channel number to a multiplexer slot and channel number address
- MCODE: Interpretation program for translating multiplexer slot and channel address to octal computer address code.

The routines used in the data reduction are:

- TDUMP: Controls and prints out rod thermocouple calculations using TCCA-A routine for linear interpolation in the Chromel-Alumel thermocouple conversion table
- HPRT: Prints out requested channels in raw data form
- HFILE: Locates a specified data file on the tape and prints its contents in volt units using an HP tape positioning routine called PTAPE

- FMU: Calculates water viscosity from temperature
- SAP: Calculates water saturation pressure corresponding to a specified temperature
- SPVG: Calculates specific volume of the water vapor at saturation conditions
- TCCC: A routine for linear interpolation in the chromel-constantan thermocouple conversion table
- TCIC: A routine for linear interpolation in the iron-constantan thermocouple conversion table
- SPVL: Calculates specific volume of liquid water at saturation conditions or subcooled conditions
- HSL: Calculates enthalpy of liquid water at saturation conditions or subcooled conditions
- SAT: Calculates water saturation temperature corresponding to a specified saturation pressure.

EXPERIMENTAL PROCEDURE

General Loop Startup and Operation Procedure

All valves in the primary loop were opened and all vent, dump, and fill valves were closed. A vacuum pump was connected to a vent line at the test section exit and the air from the primary system was evacuated until a 7 kPa pressure reading was obtained. With the vacuum pump running, the gravity fill line and the pump line to the intermediate storage tank were opened. When fill water reached the vacuum pump, the vacuum pump was stopped. When the loop pressure reached the head pressure of the storage tank the loop was vented. The loop was pressurized to 340 kPa using the makeup pump. The loop water was circulated for a period of time with part of the loop flow passing through the deionizers. When the water resistivity exceeded 10⁶ ohm-cm, the deionizers were isolated, the loop pressure was raised to 3400 kPa, and the automatic controls were set. The main heat exchanger was isolated and, using power, the loop temperature was raised. The Foxboro control valve was operated manually initially to obtain the required temperature.

Critical Heat Flux (CHF)

Critical heat flux experiments were performed by maintaining the following system conditions stable: test section outlet pressure, inlet temperature, and mass flow rate. The power was then increased gradually until a temperature excursion, judged to be indicative of the CHF phenomenon, was observed with one of the rod thermocouples, at which time the power was reduced. During the approach to CHF the CCDAS was used to

scan all data parameters. The set of measurements immediately prior to the rod temperature excursion were used as the CHF conditions. If for any reason a scan could not be carried out, the point was repeated. After the power was reduced, the data reduction option of the on-line computer program was exercised. The power was then reset to a level below the CHF power and inlet test conditions for the next run were set up.

Test Section Pressure Drop

Incremental and total test section pressure drops were measured in all tests. In addition, pressure drop measurements were carried out with zero power input to the test section and the loop at 300 K.

Subchannel Coolant Temperature

Subchannel coolant temperature measurements were made with the system conditions stabilized at predetermined values of power, system pressure, inlet temperature and mass flow rate. When stable operation was obtained at the desired conditions, the CCDAS was used to scan and store the output from subchannel coolant temperature sensors.

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