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NUCLEAR REGULATORY COMMISSION

ADVISORY COMMITTEE ON REACTOR SAFEGUARDS

IN THE MATTER OF:

SUBCOMMITTEE MEETING

on

EMERGENCY CORE COOLING SYSTEM

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Date - Tuesday, 28 August 1979

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1 UNITED STATES OF AMERICA
2 NUCLEAR REGULATORY COMMISSION
3 ADVISORY COMMITTEE ON REACTOR SAFEGUARDS
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6 SUBCOMMITTEE MEETING

7 on

8 EMERGENCY CORE COOLING SYSTEM
9 - - -

10 Westbank Motel Coffee Shop,
11 475 River Parkway
12 Idaho Falls, Idaho

13 Tuesday, 28 August 1979

14 The ACRS Subcommittee on Emergency Core Cooling System met,
15 pursuant to adjournment, at 8:00 a.m., Dr. Milton S. Plesset,
16 chairman of the subcommittee, presiding.

17 PRESENT:

18 DR. MILTON S. PLESSET, Chairman of the Subcommittee

19 MR. WILLIAM M. MATHIS, Member
20
21
22
23
24

P R O C E E D I N G S

1
2 DR. PLESSET : Let's reconvene. We have one presenta-
3 tion from yesterday that I would like to go to immediately.
4 That is the RELAP 5 simulation of the Marviken test.

5 Mr. Trapp.

6 MR. TRAPP: Thank you. You heard most of the results
7 from RELAP yesterday; but there is one test result that we have
8 run. We ran the first Marviken almost a year ago and got
9 results recently. We would like to go over those test results.

10 Marviken is unique in one sense in that it is a full-
11 scale experiment. The test itself, Test 4 and 24, are blowdowns
12 through a half-meter nozzle. It's a good-sized nozzle. We were
13 mostly interested and learned most from this problem in the area
14 of subcooled choking. We used this problem to help us under-
15 stand that phenomena, to help us to develop critical flow models
16 for the subcooled flow regions. They are very important in the
17 small-break analysis.

18 (Slide.)

19 Basically, Marviken is a vessel with a discharge pipe
20 at the bottom through which the blowdown takes place. There's
21 saturated water at the top of the vessel which turns into, as the
22 vessel key pressurizes, a two-phased mixture, which result in a
23 low quality choking at the ends of these tests. Then there is
24 about 30 degrees subcooling at the bottom of the tank which
25 results in the initial subcooled critical flow criterion in the

2
1 early parts of the run, in the early parts of the tests.³⁰⁴

2 (Slide.)

3 I show this slide only to show that the nodalization is
4 uniform throughout the pipe. Basically in meter cells, a meter
5 in length, both in the tank and in the discharge pipe, and in
6 the nozzle itself. The run times for our Marviken runs, the
7 test results are about 50 seconds to the emptying of the vessel.
8 Our computer runs are anywhere from a minute to a minute-and-a
9 half on the CDC 7600.

10 I'll just show you some results from Test 4 first.
11 That is a test with a nozzle with an L/D of 23. Test 4 has a
12 larger nozzle. This result shows both the code calculations and
13 then the data on the overlay.

14 (Slide.)

15 You can see a little bit of nonequilibrium which the
16 code follows that took place in the experiment. That's really
17 not significant when it comes to the critical flow behavior, but
18 it is something that is modeled in the code calculations.

19 This slide, the next slide, shows the flow rates as a
20 function of time, first for the calculations. This shows
21 dramatically or at least visually the two regions of operation
22 of the nozzle that are in this test. Initially it's subcooled,
23 critical flow in the first part of the test, where the velocities
24 are very high. As it depressurizes, it goes through some
25 transition region. The latter part of the test is basically

1 low quality, two-phased flow. Then an overlay of the data on
2 top of the code calculations. The data is actually measured in
3 two different ways, resulting in the two experimental curves.

4 (Slide.)

5 I would like to take a moment to show you why we
6 went to a choking criterion instead of using a fine nodaliza-
7 tion, and then take a few moments to explain the actual
8 critical flow model we used in this calculation.

9 This slide shows the sound speed which translate
10 into critical flow speeds for different equation sets. They
11 were done by a characteristic analysis looking at the derivative
12 terms in different equation sets.

13 The first is a standard two-fluid model. Analysis
14 is taken out of Wallis where he calculates the sound speeds in
15 a two-fluid mixture. The sound speed varies as quality varies
16 or vapor fraction varies between the pure phase gas sound
17 speed and the pure phase liquid sound speed. That's what he
18 calls a stratified sound speed. The point to any in this two-
19 fluid model sound speed is that it's very much higher than what
20 is actually observed in experimental results which is somewhere
21 near the equilibrium results.

22 That's the result that comes out of all the two-fluid
23 flow model analyses.

24 The next one down here, I have indicated that by
25 "homogeneous."

cs4 1 If you want to analyze the sound speed in a system
2 of differential equations, it's basically the derivative terms
3 that enter into the characteristic analysis that determine
4 what the sound speed or what the choke velocity will be. This
5 system of equations has been modified by adding a rate deriva-
6 tive or an inertial drag, a drag between the phases that depend
7 upon the relative acceleration. That changes the time deriva-
8 tives and space derivatives in the system of differential
9 equations. That enters into the characteristic analysis and
10 changes the sonic speeds.

11 With that kind of term added, with a fairly large
12 coefficient for that relative inertial drag, the sound speed
13 becomes greatly depressed and comes more in line with the data
14 as far as order of magnitude.

15 The last system of differential equations that's
16 analyzed here is one in which the mass transfer rate is assumed
17 to be the equilibrium mass transfer rate which again depends
18 upon derivatives. The equilibrium mass transfer depending
19 upon time rate of change of quality or pressure, depending
20 upon how you express it. Those terms result in again another
21 increment in the depression of the sound speed.

22 It turns out in our equations, the two-fluid model
23 equation, we have all the terms that are involved in this
24 stratified sound speed as basic convective, time derivative
25 terms. But the additional derivative terms that result in a

1 lowering of this sound speed which brings it more in line with
2 the data are derivative terms that appear in the interphase
3 drag and derivative terms that appear in the mass transfer.

4 We felt because of the state of the art that there's
5 not very many people that have tried to find correlations for
6 the derivative terms in the interphase drag. There are not
7 many mass transfer models that depend upon the derivative
8 terms. They are mostly relaxation times.

9 Those models for interphase drag and mass transfer,
10 because they don't involve derivatives, would give an analysis
11 of a stratified type of sound speed. If our numerical scheme
12 is gridded up to follow those equations, we would also predict
13 such a sound speed.

14 For that reason, we decided it would be better to
15 go back and look at the choking phenomena separately from our
16 basic equations and then impose a choking criterion that takes
17 into account the mass transfer and the inertial drag just in
18 that choking model.

19 We added an inertial drag and ran it with an
20 equilibrium and used a small cell size. I think it was 1×10^{-5}
21 for the ΔX . That did predict a sound speed within about
22 1 percent of each one of these curves at a particular void
23 fraction. The analysis is thus borne out by the code results.

24 The next one shows the idealized picture we have.
25 I am going to just go over that this morning.

1 (Slide.)

2 We looked for a start at an idealized flow, a homo-
3 geneous equilibrium flow in the subcooled mode to see what was
4 happening to give us guidance for our model. This idealized
5 model is basically what was used to generate those Marviken
6 Test 4 results.

7 Here we have a nozzle. Assume it has a high upstream
8 pressure in this direction, water flowing through the nozzle.
9 We begin to lower the downstream pressure to see what happens.
10 In this stream, with a high downstream pressure, it flows water
11 through the nozzle. As you begin to lower the downstream
12 pressure, eventually a point is reached where the saturation
13 pressures reach the throat. As we try to lower the downstream
14 pressure further, there will be flashing in this idealized case
15 that will take place at the throat; and further lowering of the
16 downstream pressure essentially fills up the diffuser section.

17 The picture we have at that time, which is the
18 operation in the subcooled critical flow mode, looks something
19 like this.

20 (Slide.)

21 Upstream, there's water that has a high sound speed
22 so it's not choked by a Mach number 1 in the water but is
23 governed by the new type equation, a momentum equation. Satur-
24 ation pressures reach the throat. The velocity at the throat
25 can be calculated from the upstream pressure and the saturation

cs7
1 pressure.

2 The velocity at the throat is really much higher
3 than the sound speed, the homogeneous equilibrium sound speed
4 that exists just downstream of that throat section. The two-
5 phased mixture has a low sound speed, as seen by the flat part
6 of that flow rate in those tests at the end of the run;
7 whereas, during the subcooled part where the flow is higher,
8 the velocity is actually higher than the two-phased sound
9 speed so that the flow in the downstream part of the nozzle is
10 actually supersonic. It is really, in this idealized case --
11 there is no place where the Mach numbers equal 1.

12 This is basically the criterion that we use. We
13 calculated a saturation pressure at the throat, calculated
14 from Bernoulli's equation of velocity. We use that as the
15 critical flow until, as we depressurize, the velocity begins to
16 slow down. Eventually this velocity equals the two-phase
17 sound speed at the throat.

18 At that time we have a transition. From now on if
19 the pressure in the tank becomes any lower, there will be a
20 two-phased mixture, not only at the throat but back upstream.
21 We can use our regular two-phased critical flow criterion,
22 which is not quite the homogeneous equilibrium sound speed.
23 We analyzed a set of differential equations which we believe
24 have correct models for mass transfer and inertial drag and
25 analyzed a sound speed from that mixture which we impose in the

1 two-phase region.

2 Those results, like I said, were run with this
3 idealized nozzle, choking criterion. We found just by an
4 experimental adjustment that we got a little better result
5 with the experiment if we used a 5 percent area reduction,
6 which was done in the case of those slides.

7 After we had done Test 4, about a year ago, we felt
8 very encouraged. We were quite excited because the results
9 were quite good. We were asked to do a blind calculation on
10 Test 24, I guess about a month ago. When we ran our result, we
11 did the same thing we did on Test 4 as far as the critical flow
12 model and got about 10 or 15 percent error in our mass flow
13 rates which caused us to go back and try to investigate exactly
14 what was going on.

15 One thing we did right away, we looked at our results
16 and thought, "Let's see if we can make one adjustment." We
17 will make that based on the fact we like the flow rate to become
18 higher.

19 We thought if there's some nonequilibrium in the
20 throat field, then the throat pressure would be less than the
21 saturation pressure indicated by the nonequilibrium. We made
22 one run with just the throat pressure some percentage of
23 saturation. We used .87 percent of saturation to give us an
24 indication of the nonequilibrium.

25 We made a second run to see if we were in the ballpark.

1 These results are from that case.

2 (Slide.)

3 This was on Test 24 with a short nozzle. There was
4 the pressure in the tank at the bottom. There was the data.
5 The same kind of phenomena. We were close but not as close as
6 we were on Test 4.

7 The next slide shows the flow rates for that same
8 Test 24.

9 (Slide.)

10 Both the code calculations showing the subcooled
11 region and the two-phased region, then the data for that
12 problem.

13 - (Slide.)

14 Because we saw a difference in the modeling of these
15 two tests, one of them using essentially equilibrium and the
16 other to get reasonable results, we used a nonequilibrium
17 throat pressure that was 7/8 of the saturation pressure.

18 MR. THEOFANOUS: Excuse me. Did you do that all
19 through the calculation?

20 MR. TRAPP: Only in the subcooled part. The things
21 I am trying to emphasize relate to our subcooled model right
22 now.

23 This caused us to go back and say, "Let's look at
24 the data and see if we can figure out a little more of what is
25 happening."

cs10 1 We took a preliminary look at some of the data.
2 This is what we came up with, again looking at the subcooled
3 region, basically at 10 seconds. We picked 10 seconds to
4 make our first look at the data for Test 4 and Test 24. We
5 also did it for Test 22, which is on the chart, and did it for
6 four other tests. We did it for all the nozzles that have a
7 half-meter diameter and a nozzle with a .3 diameter.

8 We noticed this. All the results on the slide are
9 just the data. Saturation pressure -- look at Test 4, in
10 particular. Saturation pressure at the nozzle; the throat
11 pressure measured in the experiment. For Test 4 the ratio of
12 the saturation pressure to the throat pressure. There is
13 about .96 parts of equilibrium. The throat pressure was 4
14 percentage points down from saturation.

15 Then we could take the throat pressure in the experi-
16 ment and the pressure at the bottom of the tank and from
17 Bernoulli's equation calculate the velocity that would exist
18 at the throat.

19 Taking that velocity and the actual measured flow
20 rate, we can figure out how much of the pipe must have been
21 flowing full and get an area reduction. For Test 4 it came
22 out to .84, a 2-D effect for air conduction.

23 We did the same for a number of tests. In all the
24 tests, we basically noticed, in the subcooled region, about a
25 15 percent area reduction.

1 The worst case was Test 24.

2 There appears to be about a 15 percent vena contracta
3 effect even in the Marviken nozzle.

4 It turns out the equilibrium is about .9 in every
5 test except this Test 24, which has a very short nozzle. In
6 fact, .96 was the highest one out of the six tests we ran the
7 data through for.

8 We feel we may have a little better understanding
9 that there is some nonequilibrium going on in the Marviken
10 tests, especially in the short nozzle that was chopping off.
11 Apparently there is a fairly consistent vena contracta effect in
12 all the tests. We intend to go back and run some of these tests
13 over, using these parameters in our critical flow model, to see
14 if we have a correct understanding of what was going on in
15 those tests.

16 All the tests show something, but Test 4 gives us
17 another problem that comes up in all of these critical flow or
18 experimental results. We ran this Test 4 with -- we ran it
19 with PSAT. We began to ask how did we get such good results
20 when it doesn't compare with the data?

21 It turns out this nonequilibrium effect and this
22 area effect are in opposite directions. If you take this
23 idealized PSAT flowing area as a reference case, when the
24 throat pressure is less than the saturation pressure at the
25 throat, that causes the velocity to be faster at the throat;

cs13
1 right?

2 So that would be more mass flow rate than the
3 idealized case. Because there is a vena contracta effect, that
4 sees less area. It goes back in the direction of the idealized
5 case.

6 We said, "Let's assume we had a PSAT here in the
7 data and calculate what the data's area reduction would be."
8 It turned out to be .94. We ran with a .95 on the first test.

9 It shows we can get reasonable results. We have to
10 be very careful. We might be not modeling, what we think we
11 are modeling. We might be mixing up effects.

12 (Slide.)

13 One of the conclusions I have from our analysis is
14 that for experimental results it would be helpful to have
15 extremely smooth nozzles to eliminate geometric effects.
16 That way we could have a better handle on what is really going
17 on in the actual critical flow process. It would give us
18 better control over the flow rate of the depressurization rate,
19 which is the thing that is being controlled in those experi-
20 ments.

21 We also saw, in just a summary, that a critical flow
22 model must be used in these codes unless one is willing to
23 develop the correlations, the time-dependent -- not time-
24 dependent -- rate-dependent correlations for interphase and
25 mass transfer.

1 Two of the effects do complicate the pictures. Some-
2 times they are hard to sort out. If we hadn't had pressure
3 measurements in the throat of the nozzle, we could not have
4 sorted out what was due to the nonequilibrium and what was due to
5 the two-dimensional effects. The Marviken tests do show some
6 nonequilibrium, as indicated by those numbers in that data
7 analysis.

8 Thank you.

9 DR. PLESSET : Thank you, Mr. Trapp.

10 Any comments or questions?

11 MR. THEOFANOUS: What is the critical pressure ratio
12 in these tests?

13 MR. TRAPP: Vic, do you remember that?

14 MR. RANSOM: What do you mean?

15 MR. TRAPP: You mean in the entrance to the throat?

16 MR. THEOFANOUS: Yes.

17 MR. TRAPP: I think this chopped-off one was some-
18 thing like .6.

19 MR. THEOFANOUS: .6?

20 MR. TRAPP: Wasn't it?

21 MR. RANSOM: I guess I am confused.

22 MR. THEOFANOUS: Critical pressure ratio.

23 MR. RANSOM: The actual measured value was about .6,
24 yes.

25 MR. THEOFANOUS: In the subcooled region.

1 MR. RANSOM: That's relative to the saturation
2 pressure.

3 MR. THEOFANOUS: No. Critical pressure ratio just
4 compared to the stagnation pressure.

5 MR. TRAPP: Back in the tank?

6 MR. THEOFANOUS: Yes, back in the tank.

7 MR. TRAPP: I think that is reasonable, .6. Some of
8 them went down to -- I think the chopped-off nozzle went down
9 to .4. I am sure it did. I can remember the graph now. The
10 minimum is around .4.

11 DR. FABIC: The pressure taps in the exit, they are
12 not that close together, to each other?

13 MR. TRAPP: No. Farther down --

14 DR. FABIC: But you see what you call the throat
15 pressure is right at the end of that. In fact, only the latest
16 nozzles have an abrupt ending at the end. Others had steps.
17 It complicates the picture much, much more. I don't want to go
18 into that one.

19 The later nozzles have an abrupt ending. The
20 pressure traps near the end are not that close together. Yet,
21 we see that there is a very steep pressure flow line near the
22 end. I don't think we have knowledge of the throat pressure.
23 You can't claim much about the mechanistic models based on that
24 knowledge.

25 MR. THEOFANOUS: How about your calculated pressure

1 ratio.

2 MR. TRAPP: Can I say something? The longer nozzles,
3 like in the Test 4 with an L/D of 3, there were measurements
4 at the throat. The measurements Stan was talking about are
5 three measurements downstream.

6 Even there, to get the throat pressures, the taps are
7 not that close and we just took the minimum one. They do
8 show --

9 MR. THEOFANOUS: What would you be calculating
10 ideally based on your model?

11 MR. TRAPP: In Test 4, we calculated about the same
12 as the data in our idealized case. There were only like nine --

13 MR. THEOFANOUS: I mean if you took the pressure that
14 your model calculates right at the throat?

15 DR. FABIC: I am confused as to what he calls the
16 throat.

17 MR. TRAPP: We took the minimum -- they have some
18 pressure taps all the way along the discharge pipe with a
19 larger number right at the exit; but up near the entrance to
20 the nozzle, there's three or four taps. We took the first
21 one, which really turns out to be the minimum one. If you plot
22 a pressure line down the pipe, taking all the taps, the ends,
23 the ones near the throat, the minimum always appears at the
24 first pressure tap. We took that one.

25 DR. FABIC: The minimum is right at the end of the

1 pipe?

2 MR. TRAPP: Okay. There is a minimum and it goes
3 up and rapidly falls.

4 DR. FABIC: That is just the vena contracta effect?

5 MR. TRAPP: That's what I was calling the throat.

6 MR. RANSOM: The minimum pressure point -- John,
7 maybe you can clarify that --

8 (Simultaneous discussion.)

9 MR. TRAPP: There are questions whether it's choking
10 there or at the end. It has to do -- that's a complicated
11 picture, like Stan says. If you assume there was choking at
12 the end, where it was flowing full, then you'd have different
13 parameters. These parameters --

14 MR. THEOFANOUS: With this understanding, can I ask
15 you again to please tell me in very few words what is the pre-
16 scription that you used in your critical flow calculation?

17 MR. TRAPP: We assume at the throat section, which is
18 the section which we had as the first -- I can show you what we
19 call the throat section on the slide in the numerical model.

20 If we look at our nozzle, we really had the first
21 section -- one at the very exit and one in the middle of the
22 discharge nozzle. At this section here, our choking criterion,
23 we applied the minimum section where the nozzle began. At that
24 point, we imposed in the idealized case for Test 4 -- satura-
25 tion pressure existed at that throat, and that gave us

1 essentially momentum equation.

2 MR. THEOFANOUS: You just used only --

3 MR. TRAPP: Or in the later runs, we had some ratio
4 of the saturation pressure. That was the value used to get a
5 flow, a critical flow, in the subcooled regions.

6 MR. THEOFANOUS: I see.

7 MR. TRAPP: When that value -- when the two-phased
8 criterion -- you notice, if there was some quality at that
9 point, when that two-phased sound speed was larger than that,
10 we transitioned to a two-phase critical flow model which had a
11 different criterion.

12 MR. ZALOUDEK: What did this discharge pipe discharge
13 into? Was it a free discharge?

14 MR. TRAPP: Essentially free. As I remember, there
15 was a deflecting plate.

16 MR. ZALOUDEK: Then do you feel that the flow was
17 attached to the walls or detached from the walls at the vena
18 contracta?

19 MR. TRAPP: In this region here or out here at the
20 exit?

21 MR. ZALOUDEK: In the pipe. In the vena contracta
22 within the pipe?

23 MR. TRAPP: We felt from the data it indicated that
24 some kinds of separation, maybe some cavitation at the throat,
25 as it flashed. It wasn't flowing full at the throat.

cs17

1 MR. ZALOUDEK: Since it wasn't flowing full, the
2 critical flow situation should have been at the vena contracta
3 or the last place where it attached to the walls, because if it's
4 not attached to the walls, of course, it can't choke.

5 MR. RANSOM: John, could I make a comment on that?
6 This is something I feel very strongly about. I have been
7 urging people to run these critical flow tests with idealized
8 nozzles at least for a reference case so we could separate
9 some of these geometric effects from separation and/or cavitation
10 which may be occurring at the point of vena contracta. We can
11 put together a picture; even if it's separated, it could still
12 be choked in that the vapor flow in the separated region or
13 cavitated region could be supersonic.

14 Actually, from the data, looking at tests like this,
15 you cannot separate out which of these kinds of phenomena are
16 occurring. I think John will agree with me from these looks
17 that he's taken of a number of tests, an idealized nozzle, an
18 ASE bell-mouth nozzle with a smooth entrance, 10 degree entrance
19 or so, would eliminate at least one of these variables which
20 would allow the critical flow model to infer more about what
21 is happening in terms of nonequilibrium and the choking
22 phenomena, or choking criterion.

23 I don't see why they don't use this kind of a nozzle,
24 especially when the only objective of the test is to set a flow
25 rate for a given depressurization rate.

19 1 I understand that Marviken wanted to model an actual
2 nozzle inlet like you might have on a reactor vessel. That's a
3 reasonable objective. But they should have included at least
4 one ideal nozzle to eliminate and separate the geometric
5 effects.

6 DR. PLESSET : I am going to let Stan have the last
7 word. We have to move on.

8 DR. FABIC: These large tests that we get were not
9 meant to be model development tests.

10 Why use those to develop models? Why should they
11 check these models for the effect of scale using test data
12 from Marviken? I think it is wrong to start developing basic
13 models based on tests like this. I think that's wrong.

14 That's just -- what you call it -- trying to fit
15 the model to agree with the data.

16 DR. PLESSET : All right.

17 MR. THEOFANOUS: Could I ask you on the basis of
18 what you learned from that, what do you recommend to use for a
19 critical flow for a reactor calculation?

20 MR. TRAPP: We feel good about using this model,
21 this idealized case, as a reference case. We would like to be
22 able to quantify it a little better. Some departures from
23 nonequilibrium at the throat so we have the capability of
24 putting in some throat pressure that is different from
25 saturation.

1 MR. THEOFANOUS: So at this point you don't have
2 anything to recommend specifically?

3 MR. TRAPP: Other than that idealized model with
4 some paramaters to take into account the nonequilibrium.

5 Thank you.

6 DR. PLESSET : Thank you.

7 Our next presentation is on the BEACON code.

8 MR. BROADUS: Good morning. I am Chuck Broadus. I
9 am responsible for the BEACON code development program here at
10 INEL. This morning, in discussing the BEACON program, I am
11 going to be covering the purpose and philosophy behind BEACON,
12 as well as some of the methods used in the BEACON model develop-
13 ment. I will present then what our current released version of
14 the code consists of, that is, BEACON Mod 2, and some of the
15 results we obtained by running this version of the code.

16 Finally, I will wind up with a summary of the current
17 work that is ongoing in the BEACON code.

18 (Slide.)

19 DR. PLESSET : Any condensation to save time would
20 be appreciated.

21 MR. BROADUS: Okay. The purpose of the BEACON code
22 is to provide a best-estimate containment analysis capability
23 in order to evaluate the actual transient or phenomena
24 associated with a reactor containment.

25 To discuss the reasons this is important, I want to

cs21 1 cover what the BEACON philosophy is. To put it into perspec-
2 tive, I would first like to cover what the current licensing
3 philosophy is for reactor containments in the United States.

4 (Slide.)

5 The current licensing philosophy in the United
6 States consists of using conservative codes with conservative
7 input to come up with a conservative analysis.

8 What this does, this produces a "conservative result,"
9 which is more than adequate for the licensing of nuclear
10 power plants. However, it does have some drawbacks.

11 First of all, we do not know the degree of con-
12 servatism involved in the analysis. We know that it is con-
13 servative, but we don't know the safety factor involved.

14 Second, the use of conservative or homogeneous-type
15 codes may cover up important problems or phenomena.

16 Finally, for some analyses, high pressures and
17 temperatures are not conservative. For some analyses you want
18 low pressures and temperatures; and in this case, if you have
19 a code that by its nature calculates conservatively high
20 pressures, all you can do is try to go for conservatively low
21 input data. Then, again, you don't know where you stand as
22 far as the margins go.

23 (Slide.)

24 Comparing this to the best estimate or BEACON
25 philosophy, we use a best-estimate model in order to predict the

ce-22
1 actual transient phenomena. That is, we are trying to include
2 in the best-estimate code the actual physics of the problem or
3 at least as much of the physics of the problem as it's possible
4 to do.

5 What this buys us is that we will have a benchmark or
6 a basis for comparison with the licensing codes so that we can
7 determine what our safety factor is.

8 Secondly, if we have done our homework and have in-
9 cluded enough of the physics of the problem, it should show up
10 all of the problems -- potential problems -- and phenomena
11 that would be associated with a given analysis.

12 Also, the best-estimate or BEACON type of code may be
13 used for European licensing type of analyses where you calcu-
14 late a best-estimate analysis, and then either add or subtract
15 a safety factor, depending on if you need a high pressure to
16 be conservative or a low pressure to be conservative.

17 Finally, we feel that BEACON is a general enough
18 tool as far as the geometry, the inclusion of the physics of
19 the problem, that it can be used to handle further problems
20 that may crop up in the containment analysis area. There have
21 been a number of problems that have occurred in the past where
22 BEACON would be a good general tool that could maybe not
23 handle the problem completely in an ideal way but could at
24 least get a handle on the order of magnitude of the problem.

25 (Slide.)

CF 3
1 I will now move on to the methods used in the
2 development of BEACON. In general, we tried to use suitable
3 existing models where this has been possible. These models
4 include at the heart of the BEACON program the LASL KFIX
5 numerical scheme, which is a two-dimensional, two-phased, non-
6 equilibrium type of a calculation.

7 We added to that the INEL heat conduction subcode to
8 account for our wall and structure heat transfer.

9 We have two water properties or equation of state
10 packages, one which is a quick-running ideal-properties
11 package based on a Los Alamos package; and the other is a
12 slower-running ideal-properties package based on the Brookhaven
13 National Lab.

14 We also included the INEL dynamic storage routines
15 in order to conserve the core usage for a given problem and
16 the INEL intraprocessing and plot packages.

17 (Slide.)

18 However, in cases where suitable models were not
19 available, we have developed our own. These included the use
20 of the addition of an air pump on an end to the KFIX numerical
21 scheme. This is necessary in a containment code.

22 We added a mass momentum and energy source model into
23 BEACON. This is primarily to model break locations, either
24 directed or nondirected. However, the same source model can
25 be used to model such things as containment sprays, fans, and

24
1 coolers.

2 We added a wall film model to track and calculate the
3 mass of film that becomes de-entrained from the bulk fluid flow
4 onto the walls or may condense on the walls.

5 We have developed a set of heat transfer correla-
6 tions which are based on the physics of the problem and
7 utilize the unique information which BEACON provides. This
8 includes localized velocities, the localized thermodynamic
9 properties, as well as the film thickness which would tend to
10 modify the heat transfer coefficient.

11 We also added a number of options which allow us to
12 model a general geometric -- properties that are found in most
13 nuclear power plants. These include a generalized mesh coupling
14 scheme whereby we can connect a number of meshes in a very
15 generalized way. We have a variable mesh spacing option
16 where we can mesh very finely in a certain area, where we are
17 interested in some phenomena that we are expecting there; as
18 we move away from that area, increase the cell dimension.

19 We have added a partial flow blockage capability
20 whereby we can partially block off a cell and add a loss
21 coefficient to account for a small piece of equipment or pipe
22 running through that cell.

23 We added a lump parameter region model in order to
24 model large rooms we are not particularly interested in the
25 variation of thermodynamic properties; and finally, we added a

25 1 restart capability so that expensive codes can be run a little
2 bit at a time, or expensive problems can be run a little bit
3 at a time so that you don't waste a lot of money on a problem
4 that's going to bomb.

5 Finally, all of these items were included in modular
6 fashion so as future models come along, better models, it will
7 be very easy to remove the current model from the BEACON code
8 and introduce the improved model.

9 All of these models have been incorporated into
10 BEACON and are included in the currently released version,
11 BEACON MOD 2-A. BEACON MOD 2-A is a two-dimensional, non-
12 equilibrium model capable of handling two components, air and
13 water; and two phases, the air vapor phase, as well as the
14 liquid phase.

15 It can handle complex modeling or complex geometries.
16 It is good for the short to the intermediate term; and it has
17 heat transfer capabilities from structures to individual cells
18 or group of cells; and it can also track a film that may be
19 covering those heat structures and modifying the heat transfer
20 coefficient.

21 (Slide.)

22 Now I would like to present some of the results that
23 we have obtained using the released version of BEACON MOD 2-A.
24 Specifically, I would like to cover the Battelle-Frankfurt D-15
25 test which was the basis of a CASP or containment analysis

26 1 standard problem which we participated in in Frankfurt, Germany.

2 This consisted of six rooms. Rooms 6 and 8 were
3 connected by a duct which passed through Room 4. The other
4 rooms were all connected by way of orifices.

5 The break location was up in Room 6 in the corner;
6 and it consisted of steam. This was originally intended to be
7 a blind problem. We did participate in that. The results I
8 am going to show you today are not the blind problem results
9 that we turned in.

10 At the time that we ran, we had to turn in the blind
11 problem results, we had just integrated the heat transfer model
12 and the wall film model into the BEACON code. As a consequence,
13 we still had a number of coding problems which caused an over-
14 prediction of the temperatures in the compartments. This
15 primarily had to do with the interaction between the wall film
16 and the cell. We also had oscillations which tend to obscure
17 the results. The oscillations, it turned out, were caused
18 by Modeling Rooms 7, 4, 5, and 9, two rooms away from the break
19 as lumped parameter regions connected by lossless orifices.

20 At this time in the MOD 2-A version of the code we
21 do not have the capability of modeling irreversible losses.
22 This set up oscillations between the lumped parameter regions
23 which carried back into Rooms 6 and 8 which were modeled as
24 two-dimensional mesh regions.

25 The results I will show you today use the exact same

cs27
1 data that we used for the initial submittal. However, the
2 rooms away from the break were also modeled as two-dimensional
3 regions. Those are the only changes in the data.

4 (Slide.)

5 This is a comparison of the BEACON calculations for
6 the pressure in Room 6 shown as the red dashed line with the
7 measured data in that same room.

8 In Room 6 we overpredicted the pressure transient;
9 and in Room 8, in this room we underpredicted the pressure
10 transient. The reason for this is that, as I mentioned before,
11 in this version of the code, we do not have the capability of
12 modeling loss coefficients or some kind of an irreversible loss.
13 We attempted to account for this loss or this effect by re-
14 ducing the area.

15 This worked well for all the orifices in the problem;
16 but evidently, for that duct between Room 6 and 8, we overdid
17 it, causing the overshoot of pressure in Room 6 and the under-
18 prediction in Room 8.

19 In all of the other rooms away from the break, we
20 had an excellent agreement with the data.

21 (Slide.)

22 This is a comparison of the temperature data in Room 6
23 for three different thermocouple locations. The first two
24 lines represent the thermocouple location closest to the break;
25 the next two, a little farther away; and finally, the last two

1 curves represent thermocouple location located near the duct
2 entrance into Room 8.

3 A number of things can be seen from this graph.
4 First of all, just the fact that BEACON is calculating three
5 different thermocouple locations points out its ability to
6 calculate a nonhomogeneous two-phase type of a problem. We
7 were the only one that participated in the CASP problems either
8 of the Americans or Germans which were able to show up this
9 capability.

10 The next thing were oscillations in the temperature
11 curves for BEACON. This is due directly to using too large of
12 a transfer for the heat calculations. If we reduce the time
13 step to where it's more on the order of the fluid calculations
14 time step, then those oscillations disappear.

15 Next, I want you to notice on the slide the delay
16 that occurs right here, or the overprediction of BEACON as
17 compared to the data during the early transient. I will cover
18 that later on a different slide.

19 The last thing here is the underprediction or the
20 apparent underprediction of the data of the thermocouple away
21 from the break.

22 This is due to the fact that BEACON cannot calculate
23 a steam front passing through the room. You can see the
24 effects of the steam front as it hits the thermocouple at this
25 point where we get a rapid rise in the thermocouple temperature.

29 1 BEACON, because of its nature -- it's a finite
2 difference code -- tends to diffuse the front over a number of
3 cells. The air fraction, which has a significant effect on
4 that thermocouple, or on the temperature calculated by BEACON,
5 is not purged out of that cell as rapidly as it is in the
6 regular problem.

7 If you look at a thermocouple in these rooms as a
8 heat sink, a small but finite heat sink that is initially cold,
9 the first steam that is going to come through there will treat
10 the thermocouple as any other heat sink. It will condense out
11 on the thermocouple.

12 You expect during the early transient for the thermo-
13 couples to be measuring more of the saturation temperature than
14 the actual temperature in the room.

15 (Slide.)

16 This is the same data as in the previous slide.
17 However, what is plotted, superimposed in red, are the
18 thermocouple location saturation temperatures as calculated
19 by BEACON. This produces much better results, and we tend to
20 converge directly onto the measurements of the thermocouples.
21 So from this, we concluded that at least during the early time
22 period after the steam has passed the thermocouple, that what
23 the thermocouple is probably measuring is the saturation
24 temperature.

25 (Slide.)

cs30 1 I will move on to the subject of the delay time or the
2 overprediction you saw in the first slide. This shows the
3 temperature transient in Room 7. This is a room away from the
4 break. That's the temperature -- the data is the blue line.
5 the BEACON-calculated gas temperature is represented by the
6 gold line on this graph.

7 As we participated in the CASP workshop in Germany,
8 we noted that most of the other codes that participated had
9 the same trend, that is, to overpredict the data during the
10 early transient. Further, it was noted that this transient
11 could just about be backed out or predicted based on the
12 pressure transient and an adiabatic pressure-type calculation.

13 When we modeled in BEACON, very crudely, a thermo-
14 couple -- crudely in that we used the proper dimensions, but we
15 had to use some average properties from what we knew of the
16 thermocouple. We put it in Room 7 and made a run. The results
17 you see are represented by the red curve. The red curve does
18 match the data much better. This is the center-line temperature
19 of the thermocouple.

20 We submitted these results to Germany; and they have
21 done some testing on their thermocouples. They were claiming
22 thermocouple delay times, response times, only on the order of
23 milliseconds; but they had calibrated them based on dunking them
24 in hot water or putting them in a steam jet. This is quite a
25 bit different from a thermocouple sitting in a relatively

c-31
1 still-air atmosphere.

2 They are now saying they don't really know what the
3 thermocouple response is in this early time period, but it looks
4 to be more like on the order of 1, 2, or 3 seconds.

5 The other thing to note here is the way BEACON
6 underpredicts the temperature transient toward the end or after
7 1-1/2 seconds. This again is when the steam front comes through
8 the room and comes in contact with the thermocouple. There
9 again BEACON has diffused the steam front and it's not going to
10 pick that up immediately. We would expect, if we ran the
11 problem out further, for the BEACON-calculated temperature to
12 eventually move up to the thermocouple temperature.

13 It is not plotted on here; however, the saturation
14 curve for BEACON for this particular cell tends to follow the
15 lower curve out to about .4 seconds; and then it makes kind of
16 a parabolic increase up to matching this data out at about this
17 point. Again that would tend to point out that what the
18 thermocouple is measuring during that part of the transient is
19 probably more saturation temperature.

20 I will now move on to the current work that is going
21 on in the BEACON program.

22 (Slide.)

23 We are currently involved in developing BEACON MOD 3.
24 This involves the addition of three major models. The first
25 model -- and by far the most involved model -- is the

cs32

1 best-estimate correlations model to govern the interphasic heat,
2 mass, and momentum transfer. Dr. Sahota will discuss that in a
3 few minutes with you.

4 The second model we added is the form and friction
5 loss model, to allow us to handle wall friction as well as
6 entrance and exit losses.

7 The third model is illustrated by the figure -- what
8 we call an out-of-plane coupling model. This increases our
9 geometric modeling ability. We connect an interior cell of
10 one mesh to an interior cell of another mesh by way of a one-
11 dimensional region.

12 (Slide.)

13 As the BEACON MOD 3 program is wound up and issued to
14 the National Energy Software Center, we will be moving, as
15 Dr. North pointed out yesterday, into a period of developmental
16 assessment where we are going to be evaluating the program.
17 The purpose of this developmental assessment portion of the
18 program is to define the range of conditions and problems to
19 which BEACON is applicable. We are going to do this primarily in
20 four steps -- four steps for each particular problem, I should
21 say.

22 The first step will be the problem setup and documen-
23 tation. That is, we are going to examine the geometry, the
24 boundary conditions, and then set up the problem to the best
25 of our ability and document why we modeled the given problem in

1 a certain way.

2 The second step will be a blind run. "Blind" here
3 should be in quotes, because we will have access to the data.
4 However, we are going to attempt to not look at the data until
5 we do get a satisfactory first run, just based on what we have
6 done with the input.

7 The third step, then, would be the data comparison
8 where we will compare the results of the blind run with the
9 data.

10 The fourth step will be our learning phase where,
11 provided the results of Step 3 are not satisfactory, we will
12 modify the data within our best-estimate limitations, that is,
- 13 such things as nodalization, in order to determine where we
14 went wrong and to learn how to better model future problems.

15 The types of problems that we are going to be inves-
16 tigating fall into two different categories. We have the
17 separate-effect-type problems which are intended to look at a
18 certain capability or phenomena that BEACON calculates. These
19 include such problems as the entrainment/de-entrainment
20 experiments which are being performed at Drexel, as well as
21 the Lahey two-dimensional flow work being done at Rensselaer.

22 The other category of problems are the integrated
23 containment problems. For these we are relying heavily on the
24 Battelle-Frankfurt C&D series tests, because we have good
25 access to the data, and the type of data that they have taken is

1 particularly suited to the BEACON developmental assessment.

2 The other test we will be looking at -- other type of
3 tests is like the CVTA -- Carolinas-Virginia Tube Reactor test --
4 to evaluate stratification-type capabilities.

5 This concludes my presentation. Do you have any
6 questions?

7 DR. PLESSET : Let me ask you one. No, I have two
8 questions. You can say yes or no.

9 Do you expect to get the capability to analyze an
10 ice condenser containment?

11 MR. BROADUS: This would be included in future work.
12 As I mentioned, we are moving on to a period of developmental
13 assessment which should last about a year. After that, it will
14 be assessed whether we should continue working on it or not.
15 Then we would consider a problem such as that.

16 DR. PLESSET : Would you be able at some time to
17 evaluate the effect of a hydrogen burn on the containment?

18 MR. BROADUS: We have not --

19 DR. PLESSET : In particular, an ice condenser plant.

20 MR. BROADUS: These are, again, future models. We
21 do not have a hydrogen tracking model in there now. It has been
22 considered for future work.

23 DR. PLESSET : Thank you. I think we will go on to the
24 next part of the presentation. Again I plea for brevity.

25 MR. SAHOTA: Thank you. I would like to mention,

cs35
1 even though we have the capability of including an inert gas --
2 I think in the type of problem, we really need an equation. It
3 may not be easy.

4 DR. PLESSET : I didn't say it would be easy. I
5 thought it might be important.

6 MR. SAHOTA: It might be important. We have talked
7 about including mass diffusion.

8 (Slide.)

9 In the best estimate, we have tried to maintain
10 reality, sometimes at the expense of -- the idea has been to
11 develop equations which apply specifically to the containment
12 applications. However, the coordinations should be general
13 enough to cover our situation. Consequently, the equations are
14 correlations, but we feel they should be reasonably activated not
15 to invalidate the core if run under the extreme situations.

16 (Slide.)

17 Mr. Broadus mentioned BEACON is a two-dimensional
18 transient, two-component, two-phase nonequilibrium code. The
19 unequal velocities between the two phases lead to interphasic
20 drag.

21 The second thing is the thermodynamic nonequilibrium
22 give rise to heat and mass transfer in the presence of inert
23 gas. Since that is present, the problem is further complicated.

24 (Slide.)

25 For the analysis, we considered a very simple flow

1 regime approach. The first one is the dispersed droplet-type
2 flow.

3 The second is bubbly flow. Then the flows with
4 void fraction close to .5, and then we have flows with other
5 void fractions. Actually, in your slides, this is incorrect.
6 It should read 0.5 less than 1, and 0 less than 0.5.

7 We have given rigorous treatment to the dispersed
8 droplet-type flow, feeling this is the most frequently
9 encountered type of situation in containment application.

10 However, we also have tried to handle bubbly flows
11 in a similar manner, and less emphasis has been placed on the
12 other two types of flows.

13 Basically, the approach is for dispersed droplet and
14 bubbly flows. We try to get the interphasic exchanges for a
15 single droplet and then consider the distribution of the
16 droplets or the bubbles. We interpret those rates over various
17 sizes.

18 For void fractions close to .5, since there is some
19 doubt whether bubbles in liquid or droplets in gas, we assume a
20 varied type of interphase and used some steady-state correla-
21 tions to predict the rates.

22 The other flow situations are simply hindered by the
23 combination of the above results.

24 Let's assume a droplet has dispersed in an air vapor
25 mixture. One can calculate the interphasic rates of heat, mass,

cs37 1 and momentum transfer from some standard correlations.

2 (Slide.)

3 This is the well-known equation for drag. Once the
4 drag equation is unknown -- which is basically this data -- in
5 our standard textbooks, the development for the drag equation
6 yields an expression like this.

7 All I can say is this is only dependent upon
8 Leonard's number. Once that is known, which is known from the
9 BEACON calculations, we can calculate the drag coefficient.

10 For the mass transfer in the presence --

11 MR. THEOFANOUS: This is for solid spheres?

12 DR. PLESSET : Yes.

13 MR. THEOFANOUS: Do you expect any difference for the
14 particles?

15 DR. PLESSET : Or even for the detection of the
16 possibility of flow?

17 MR. SAHOTA: Yes. As I said, we have tried to keep
18 a very simple approach. However, Washington State has done
19 some type of work. He has had reasonable success assuming
20 droplets as spherical, solid spheres.

21 However, I would like to mention that his work has
22 been primarily in the nozzles where the droplets are really
23 small droplets. I will mention about the momentum exchange
24 between the droplets and the gas phase. That is automatically
25 included in the BEACON calculations. Like if a droplet

cs38

1 evaporates and there is a momentum exchange between the
2 droplet, subphase and the gas phase, which could affect these
3 results. Those are included.

4 MR. THEOFANOUS: We are talking about the drag
5 coefficient here, not the solid effect. There's quite a bit of
6 information that indicates the drag coefficient can be quite a
7 bit higher.

8 MR. SAHOTA: Right.

9 MR. THEOFANOUS: When the drops are deformable. I
10 think the study you mention -- it is probably true. In the
11 nozzle, the particles are so small there is no --

12 MR. SAHOTA: Either the particles are so small, or --

13 MR. THEOFANOUS: That wouldn't be the case in your
14 case. The particles wouldn't be that small.

15 MR. SAHOTA: What I was really going to say is we
16 have some other large uncertainties which completely supersede
17 this unknown effect, like droplet-like distribution. We have
18 assumed a droplet-like distribution. We have established a
19 maximum droplet radius based on numbers which was sort of a
20 mechanical trade-off.

21 DR. PLESSET : Maybe you should take a global view.

22 MR. SAHOTA: Some people have done it. We take the
23 droplet approach; you take something and it applies very well
24 under certain conditions. It does not hold that well under
25 other conditions. So what you end up doing is putting in 10 or

cs30
1 15 equations to cover different types of situations. That's
2 what we tried to avoid. I don't claim they would be accurate;
3 but at the same time they should give us answers within an
4 order of magnitude and should really not vary.

5 Anyway, the mass balance in the presence of inert gas
6 gives you an expression like this, where W is the mass fraction
7 of the volume at the interface, and infinity is the ambient
8 extreme.

9 GM is the mass transfer coefficient. I would discuss
10 the calculation of GM, which is the mass transfer coefficient
11 in a moment. However, I would like to add that in the case of
12 the inert gas not being present, we use the expression given
13 by Shadl.

14 MR. THEOFANOUS: That means you consider the important
15 step to be the condensation step from the gas phase to the
16 liquid. Wouldn't it be important that the heat has to be con-
17 ducted away from the interface? Don't you think that would be
18 the limiting step, not the transition from the gas to the
19 liquid?

20 MR. SAHOTA: I don't understand.

21 MR. THEOFANOUS: It would simply mean under your
22 conditions the kinetic limitations would be unimportant and
23 the actual consideration would be lying in the conduction away
24 from the droplet interphase into the bottom of the drop.

25 MR. SAHOTA: That is what it is. We are taking the

10
1 production into droplets in the gas phase.

2 MR. THEOFANOUS: I realize that. I am saying that
3 doesn't agree with it. It would seem under your conditions
4 kinetic limitations would be unimportant. That's just the
5 opposite.

6 MR. SAHOTA: You are saying the conduction in the
7 droplet would be important?

8 MR. THEOFANOUS: More than kinetic.

9 MR. SAHOTA: Liquid is not --

10 MR. THEOFANOUS: We are not -- you say when you talk
11 of gas, you use the science equation. That tells me, then, in
12 that case you are using kinetic limitation. I am saying
13 kinetic limitation shouldn't be important and conduction should
14 be important.

15 MR. SAHOTA: I take it back. I was thinking about
16 this inert gas phase. You are right. Again, how many applica-
17 tions do we encounter in the containment where there's no
18 inert gas? What I pointed out is we are trying to maintain
19 complexity. The point is you might end up in a flow situation
20 where there is no air.

21 I just mention that we are adding that expression.
22 I don't believe that is accurate, but that is just to force
23 the computer calculation.

24 MR. THEOFANOUS: Fine. I don't want to belabor the
25 point. It seems to me, however, you want to avoid glaring

41 1 inconsistencies or glaring mistakes. I think it's an error to
2 put something like that in the calculation. It could be that
3 during an actual calculation a subcompartment is getting some
4 of the air, and what is left is not too much air. Maybe some-
5 body then that applies the correlation is not being so careful
6 and is using the wrong calculations in the place where the
7 droplets are growing, condensing, without the presence of air.

8 MR. SAHOTA: Yes.

9 MR. THEOFANOUS: That's enough.

10 MR. SAHOTA: I get your point. But I missed the
11 point; I thought you were talking about the presence of inert
12 gas. That is where our main effort has been present, in inert
13 gas and dispersed droplet type of flow.

14 (Slide.)

15 Similarly, the energy balance at the interphase gives
16 rise to expressions like these.

17 This is really valid for number one. You get more
18 expressions.

19 Also, we assumed it to take advantage of the analogue
20 in the species and energy equations.

21 GH is equal to GM, which is the mass transfer co-
22 efficient. These two can be calculated in terms of heat or
23 mass transfer coefficients in the absence of an inert gas. The
24 distribution of the species equation, if it's given in years,
25 in an expression like this, so that if one knows the heat or

1 mass transfer coefficients in the absence of inert gas, one
2 can calculate the other quantities in the presence of inert gas.

3 This GM and GH could be calculated from well-known
4 equations.

5 Once one knows the mass effect on the droplet, all one
6 has to do is use droplets. To do that we considered distribu-
7 tion on the droplet size. The maximum droplet radius -- R is
8 the dimension of the droplet. The maximum droplet radius is
9 calculated based on mechanical calculations.

10 Once the droplet sizes are known, we just took those
11 analytic expressions and interpreted it over all droplet sizes.
12 The total drag is given by this expression. The B/R is given
13 by this expression. We were able to give analytical numbers by
14 interpreting these.

15 DR. CATTON: Did you get the same kind of analytical
16 results?

17 MR. SAHOTA: That is what that is.

18 DR. CATTON: To one-half?

19 MR. SAHOTA: I might have an error there, but I
20 think this equation is correct.

21 DR. CATTON: I thought it was for flow.

22 MR. SAHOTA: That is for black leg.

23 DR. WU: Low and intermediate.

24 MR. SAHOTA: That's correct. If you really take a
25 maximum droplet size and calculate -- look at the velocity

1 between the two. You get reasonably small numbers.

2 (Slide.)

3 Allow me to talk briefly about the bubbly type of flow.
4 For the drag coefficient, again I am going to have some ob-
5 jections here. What we did was considered a single bubble and
6 used the same approach as for droplets to calculate the drag on
7 a single bubble.

8 An energy balance on the bubble yields expressions like
9 these where GH is the heat transfer coefficient in either case.
10 Ideally, one should calculate this liquid heat transfer co-
11 efficient from the distribution of transient conduction
12 equations, or, in other words, using it, and one should be able
13 to calculate the heat transfer coefficient that way.

14 However, to do that, one needs to have the age of the
15 bubble and the history, the coefficient history in the liquid
16 case which requires a tremendous amount of computer storage.

17 So again we said, okay, we would use this equation,
18 consider it quasi-steady, and calculate this heat transfer in
19 an approved manner just because of the computer storage problem.

20 Once the energy is known, again we interpreted those
21 overall bubble sizes to calculate the macroscopic exchanges.

22 (Slide.)

23 The third type of regime is void fraction which is
24 close to .5. In the absence of any other information we
25 assumed a wavy liquid-air-vapor interphase. One could calculate

1 the stress at the interphase if that were known. One could
2 also calculate a standard number based on well-known solutions,
3 based on macroscopic exchange rates.

4 Those exchanges can be calculated by multiplying by
5 the area. That is based on the existence of droplets.

6 (Slide.)

7 The other flow regimes are simply the extensions of
8 the previous flows, flow regimes. That is a void fraction of
9 .5 and 1.0. We assumed the velocity and temperature -- we
10 ignored the velocity and temperature gradients. The phenomena
11 is basically in the gas phase. Therefore, one could simply
12 calculate the drag heat and mass coefficient.

13 However, Between void fractions of 0 and .5, the
14 conduction is in the liquid phase.

15 Similar types of approaches have been used for the
16 heat and mass transfer coefficients.

17 Just to reiterate, all I really want to reemphasize
18 is that our main emphasis has been this droplet flow in a
19 containment in the presence of inert gas. However, sometimes
20 we do run into situations where sometimes, because of numerical
21 errors, a particle or cell would give you some mass fraction.
22 We need to know the flow domains. Therefore, we have added sort
23 of generalized best-estimate interphasic correlations.

24 However, there are other types of correlations
25 existing in the BEACON calculations which are meant for special

1 emphasis. If one wants, you could use those.

2 That concludes my presentation.

3 DR. PLESSET : Thank you.

4 You have a very ambitious program. It is commendable.
5 If you are successful, it would have applications in other
6 areas aside from the containment flow problems. I wish you
7 luck.

8 MR. SAHOTA: Thank you. I think we will need that.
9 I was surprised to find that it was really based on some --
10 like -- bubble formations during that depressurization. We have
11 been able to successfully use that correlation for containment
12 applications. We said if we could use that, which was really
13 based on bubble formation and bubble growth, assuming constant
14 equilibrium during the life of the bubble -- if we could use
15 that for droplet type of flows, we certainly stand a better
16 chance of using some approach like that.

17 DR. PLESSET : Thank you.

18 MR. SAHOTA: Thank you.

19 DR. PLESSET : I think we will have to have a change
20 of gears which is required for the next subject. We are going
21 to have a presentation by Dr. Rosztoczy on the NRC Office of
22 Nuclear Reactor Regulation needs.

23 I am sure their needs are very modest.

24 (Laughter.)

25 This shouldn't take long.

1 DR. ROSZTOCZY: Always.

2 (Laughter.)

3 DR. ROSZTOCZY: Mr. Chairman, as you indicated, my
4 responsibility here is to give a brief summary of some of the
5 most important licensing data needs that we have. I would
6 like to specify that I am going to talk about only those data
7 needs which relate to Semiscale or LOFT. Also, I am going
8 to talk about short-term data needs, basically programs that
9 we expect can be completed within the next year or two.

10 With those two qualifications, let me go to the
11 slides.

12 (Slide.)

13 The first area where we have some data needs is the
14 upper head injection system. As you might recall, we have
15 completed our review of the upper head injection system back in
16 1977. We presented the results of those reviews to the
17 committee at that time.

18 One of the main conclusions of the review was that
19 the available calculational techniques had very large uncer-
20 tainties associated with them. These were sufficiently large to
21 kind of mask the possible benefits or drawbacks of the UHI
22 system. Therefore, we couldn't really decide how much of a
23 benefit those UHI systems represented.

24 In order to be able to have a better understanding of
25 other systems, like UHI and possibly other systems, ECC systems,

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1 there is a definite need for more experimental information in
2 this area.

3 This experimental information should have the under-
4 standing of the UHI system performance, and it should help us
5 to develop and verify calculational methods which can then be
6 used to analyze these systems.

7 We have requested this type of work back at that time.
8 The program has been developed for this purpose. The program
9 was ready to start approximately a year ago last summer. In
10 the first test -- there were three tests -- we ran into some
11 difficulties. Some of these are known as the S-07-6 problem.
12 We will come back to that later.

13 We fully agree with the people involved in running
14 this program that some of these problems have to be resolved
15 before we can go and complete the program.

16 We are waiting for the resolution of those, but we
17 would like to emphasize that this is an outstanding program and
18 the completion of this is important and urgent to us.

19 I would like to point out that the last word on the
20 slide should be "practicable." We would like to see this
21 program completed as soon as it is practicable.

22 DR. PLESSET : What is the Westinghouse view regarding
23 the validity of the proposed tests? I am sure there has been
24 discussions with them about that. In advance of the test,
25 what do they say?

48 1 DR. ROSZTOCZY: We have had numerous discussions
2 with Westinghouse on this subject. Westinghouse people in
3 general are not completely pleased with this program. We have
4 asked them repeatedly if they can propose or if they are
5 planning to run on their own any programs which could do a
6 better job than this. Throughout the past three years, this
7 discussion has been going on. They could not come up with
8 anything better than this program.

9 We all realize that a program, short of being a
10 full-scale program, will not provide all the information that
11 one would like to have.

12 The information presently available is very sketchy
13 and very limited. This program would put us one step, one big
14 step further ahead. It would not resolve everything completely.
15 We think it will be very valuable.

16 Should something develop in the program that indi-
17 cates that certain variations on this or some changes would
18 be helpful, they will be factored into the program.

19 Final resolution, I believe, was that Westinghouse
20 is not doing pretests for the program. We will run the first
21 test series. We are going to look at the data and see if we
22 are getting meaningful data out of the test program. Provided
23 the data is meaningful, Westinghouse will do all the necessary
24 calculations, blind calculations, that are necessary.

25 DR. PLESSET : What is your feeling about the

1 calculations, I think being made with COBRA? We have seen
2 some results. What are your feelings about that?

3 DR. FABIC: I would like to update the information
4 you have.

5 DR. PLESSET : Shall we do that, Zoltan?

6 DR. ROSZTOCZY: Please, go ahead.

7 DR. FABIC: The information you saw in Los Angeles
8 showed the results of the COBRA calculations for the vessel
9 only, where the boundary conditions were given by SATAN.
10 We agree this is important to the contrasting requirements, con-
11 flicting requirements. We have since then been able to merge
12 COBRA with TRAK so that now COBRA calculates the complete
13 system. They are redoing these UHI Westinghouse plain calcula-
14 tions for the complete system, not just the vessel. So it is
15 a separate system calculation.

16 DR. PLESSET : When do you think those would be
17 available?

18 DR. FABIC: I was on vacation for a month. When I
19 come back, I will know the answer to that.

20 DR. PLESSET : It's not too far in the future?

21 DR. FABIC: They already ran the calculations for
22 the system. They are now running the Westinghouse plant on
23 installments. It's a lengthy calculation. Their access
24 computer is very limited. They use Brookhaven. It has very
25 limited access.

1 DR. PLESSET : What is your attitude toward the
2 usefulness of these calculations?

3 DR. ROSZTOCZY: I would be interested to see them.
4 I have not seen them yet. Once I have an opportunity to see
5 them, examine them, I will be very happy to comment on them.

6 DR. PLESSET : Okay. That sounds like a lawyer.

7 (Laughter.)

8 DR. ROSZTOCZY: We all learn to talk that way.

9 (Laughter.)

10 (Slide.)

11 DR. ROSZTOCZY: Another area where we do need experi-
12 mental data is the small-break loss-of-coolant accident. We
13 have had one small-break Semiscale test run back a number of
14 years ago. This was S-02-6. This was the only small-break
15 test. It was selected as a standard problem, as part of the
16 standard problem. Most of the vendors have been doing pretest
17 predictions for this test.

18 The report predicting the calculations was published
19 back in the spring of 1978, somewhat more than a year ago. The
20 conclusion from the comparisons was that there are comparatively
21 large uncertainties in the calculations. The two main un-
22 certainties that we have observed were -- one of them is in the
23 depressurization rate. Some of the calculations were far off on
24 that.

25 The other large uncertainty that we observed was in

51 1 the behavior of the calculations at the time when cold ECCS
2 water is being introduced into the system.

3 The data shows some change in the depressurization,
4 acceleration in the depressurizations, but just a relatively
5 small amount at that time. Some of the calculations just
6 depressurized very fast. Because of these reasons we asked
7 for -- there were also some problems with the data. I am not
8 sure from the details, but I believe maybe the discharge flow
9 was measured. There were problems with the data.

10 As a result of this, we have asked for additional
11 small-break tests.

12 We have also observed from the licensing order we
13 have received last summer -- related to some changes in the
14 variation model of one of the vendors -- B&W -- in the review
15 of that, we have noticed that the calculations are rather
16 sensitive to changes, the small-break calculations. There are
17 relatively minor changes, and they can produce 500 or 600
18 temperature variations in the calculations.

19 At the same time, the tests required by Appendix C
20 were applied for large breaks. When applied to small breaks,
21 they make a relatively small difference.

22 Putting all of this together led us to the request
23 for additional tests and additional small-break analysis. The
24 purpose of this would be to evaluate the uncertainties of the
25 small-break calculation and to validate the calculational

1 methods.

2 We have also requested -- this is not a standard
3 problem, but there is a definite request from the NRC that
4 each of the PWR vendors should predict this test.

5 The actual test was completed in December of 1978.
6 The calculations from the vendors are not in yet. There have
7 been other complications while the LOFT test program was going
8 on. We have been asking for pretest predictions.

9 There was a change of priorities. The LOFT test was
10 done first. The small-break test has been locked in.

11 We have sent a letter to each of the vendors urging
12 them to provide these calculations as soon as possible. We
13 asked for it by September. We are in discussions with them on
14 the actual date when the calculations will come in.

15 (Slide.)

16 More recently, due to the TMI events, there has been
17 an increased emphasis on small breaks. This is a continuation
18 of my small-break slide.

19 TMI-II brought attention to various facts in
20 connection with small breaks. One is that the very small LOCAs
21 are occurring with a higher frequency than what we previously
22 considered. In the B&W case, there have been four events within
23 30 reactor years of operation, which is a relatively high
24 frequency.

25 The other thing we learned is that the plant responses

1 to very small breaks differ from the plant responses to the
2 more common small break that has been looked at and analyzed in
3 the safety evaluation models. Very small breaks are not large
4 enough to remove the decay heat from the core. The discharge
5 cannot remove all of this heat. Because of this, it is very
6 important to have other modes of heat removal.

7 Natural circulation to the steam generators, there-
8 fore, plays a very important role in this very small break.
9 Because the heat is not removed, overpressurization is possible.
10 If the break is at certain locations, like in the pressurizer,
11 at the pressurizer level, it will not empty. There are a
12 few other complications depending upon where the reactor coolant
13 comes from.

14 In order to be able to handle these very small breaks,
15 the calculational techniques have to be carefully removed for
16 this purpose, keeping in mind these serious phenomena to see
17 if they can be handled.

18 Up to now, our observation is working with the
19 various vendors. Each of them need some changes in their
20 variation models in order to account for all of this.

21 We also found that there is not sufficient experimen-
22 tal information available. We are at the point now when we
23 think the codes are predicting the basic phenomena. Where they
24 are, we do not know yet. We do need experimental evidence.

25 We, therefore, request tests -- the word is used

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1 here, a "demonstration" test. A demonstration means that
2 during the past three months in our work with the various
3 vendors, they have predicted through the analytical techniques
4 of how the plants could respond to the small breaks. They
5 predicted for some of these breaks that there will be
6 depressurization when the various ICCS systems come into play.

7 For some small breaks, there will be a hang-up of
8 pressure close to the secondary site pressure. There will be
9 no further depressurization for the next three-year period of
10 time. For other breaks, there will be pressurization up to
11 high pressures. Other means are needed to reduce the pressure.

12 We are asking for tests to demonstrate these three
13 basic behaviors of very small breaks, the depressurization,
14 the pressure hang-up, and repressurization.

15 The test program is being planned to augment each of
16 these. They will provide at least one basic test result for
17 each of the cases.

18 We have also predicted -- we have requested reactor
19 suppliers provide pretest prediction for selected tests. The
20 tests we are asking them to provide have not been selected.
21 It appears now that it will be one or two tests which will be
22 run later this year, or late this year, or early next year.

23 This has been spelled out in the latest record, new
24 Reg 0578.

25 The small-break tests have a very high priority.

cs-55 1 We would expect to see a program ongoing on this in the very
2 near future.

3 (Slide.)

4 The third document I would like to discuss is
5 Semiscale Test S-07-6. This was run as one of the first tests
6 in the new Semiscale MOD 3 system. This test was run when
7 voiding on the downcomer occurred during the test, which kind of
8 caught our attention.

9 As a result of this multiple voiding, it was rather
10 pronounced and rather long-delayed in the quenching of the core.

11 The peak clad temperatures have not changed signifi-
12 cantly; so the calculated peak clad temperature and the actual
13 pressured peak clad temperature weren't too far apart. I don't
14 have a slide with me that shows the temperature, but the
15 difference there is not much.

16 However, there is this large difference in the
17 quench time. So the oxidation, the zirc-water reaction, because
18 of the time period, is quite different than what the code
19 calculations would show.

20 In our licensing work during the past few years,
21 since 1974, since the ECCS criteria was published, shows that
22 in some cases the calculated zirc-water reaction is very close
23 to the limit specified in the acceptance criterion. There have
24 been a few cases in the past when the plant was limited by the
25 zirc-water reaction. They ran into the zirc-water reaction

1 before and into the peak clad temperature.

2 The extended time period here could result in rela-
3 tively large uncertainties in the prediction of the zirc-water
4 reaction, like something by a factor of 2 or maybe more.

5 So, therefore, we paid quite a bit of attention to
6 this test.

7 The first question was: Can the codes predict and
8 can they calculate this test? We have tried the RELAP 4 code.
9 We have tried the TRAK code in its 1-D version.

10 The conclusion from both of those tries was that the
11 codes were unable to correctly predict the experiment.

12 Yesterday you saw some slides which were done with
13 the RELAP 5 code for this very same test. They have shown some
14 promising effect for the early portion of the transient.

15 In order to put it into proper perspective, those
16 calculations were only for the blowdown portion of this
17 transient, because the RELAP 5 code is not ready yet to be used
18 for the other version.

19 We cannot run the complete version with RELAP 5 at
20 this time. Therefore, they do not know whether RELAP 5 would
21 be predicting correctly or approximately correctly the delay in
22 the quenching of the core. Some of the code modifications which
23 were outlined yesterday will provide this capability. My own
24 estimation is that RELAP 5 would have this capability next
25 year sometime, but not earlier.

1 There was also a divergence observed in the temperatures, the
2 calculated temperatures relative to the measured temperatures of
3 the first turnaround in the temperature that is not fully under-
4 stood yet, what caused it. I think more work is needed before
5 some of this is ready.

6 The conclusion is at the present time we do not have
7 a satisfactory production of these test results by any of the
8 computer codes available to us. Right now we cannot calculate
9 this step.

10 We have requested both experimental and analytical
11 studies to decide whether this phenomena is typical for PWRs
12 and whether it is significant for the licensing type of actions,
13 licensing type of decisions.

14 We do not have in all the results of the various
15 things that are ongoing to try to resolve this problem. We
16 don't have a final answer yet. However, up to now, we have
17 learned a number of things. Those are listed under the present
18 status.

19 One of them that we have learned, and we are convinced
20 of, that this phenomena is a real phenomena and it can occur in
21 large systems just as well as in small systems.

22 DR. PLESSET : How do you base that can occur for
23 large systems? Why do you feel that?

24 DR. ROSZTOCZY: That it can occur in large systems?
25 The phenomena is basically a phenomena we all know from our

c 3 1 high school physics; and from there on it's sometimes called
2 chi-square.

3 If you get into a condition that the water is
4 saturated in the downcomer everywhere, taking into account that
5 the pressures vary in the downcomer, is saturated everywhere,
6 then if you produce a void somewhere so you have just a little
7 more heat, then you have an unstable situation.

8 In the production of the void, it reduces the
9 heat, reduces the pressure, produces more boiling, more
10 flashing, which then pushes up more water, and you throw the
11 water off on the system.

12 Hence, it is not limited in size. Those of you who
13 might have a little more time can go to Yellowstone and see
14 the action at Old Faithful.

15 DR. PLESSET : It should occur in LOFT, in other
16 words?

17 DR. ROSZTOCZY: Since it came out in Semiscale in a
18 very pronounced manner, we have gone back and asked the question:
19 Is this unique to Semiscale, or has this not been observed in
20 the other test apparatus where tests have been conducted?

21 The answer is it has been observed in every single
22 apparatus on which tests have been conducted. It has occurred
23 in the German test, in the Japanese, in LOFT, in the special
24 test done on large scale, actual scale.

25 DR. PLESSET : No doubt it occurs, Zoltan. Maybe it

1 has trivial effects. If it's trivial, you don't make a big
2 thing of it.

3 DR. ROSZTOCZY: Yes.

4 DR. PLESSET : Would you take me further on this non-
5 triviality?

6 DR. ROSZTOCZY: Yes. Let me just continue along
7 the lines of the slide. I think it will bring this up.

8 DR. PLESSET : I think Dr. Tong wants to make a
9 remark.

10 DR. TONG: May I caution you that S-07-6 test was
11 caused by a poor insulation in the downcomer. We reported
12 that, and we are modifying it. The Semiscale-downcomer inter-
13 relation is going to be repaired and repressed.

14 DR. PLESSET : Yes, I know.

15 DR. TONG: The consequence you got from S-07-6 may
16 be contributive to the poor insulation. It may be atypical.
17 There should be a caution on that interpretation from that,
18 that it may be misleading.

19 DR. PLESSET : I think that's a very good point,
20 Dr. Tong. I remember when the subcommittee heard about the
21 S-07-6. There were strong reactions to the lack of insulation
22 or effective insulation.

23 DR. TONG: Yes.

24 DR. PLESSET : I don't call it a downcomer; I call it
25 a hot pipe, which is what it is, connected to another hot pipe.

1 That is Semiscale in this context.

2 I think we should be a little less -- I don't know
3 what -- I have been careful with my literal-minded friend
4 about the significance of this for other prototypical or large
5 BWR systems.

6 Now, Zoltan is being very serious about it. I think
7 it's an interesting thing, but its pertinence may be a little
8 bit open. It could be a difference of opinion. I see Zoltan
9 doesn't look happy.

10 DR. ROSZTOCZY: No. I think I agree with most of
11 those remarks. I will come back to those at the end.

12 One thing -- in our present status we are saying the
13 phenomena is real. Whether it would occur on a PWR, we don't
14 know it yet; but we don't think it's something limited to
15 small sizes.

16 DR. PLESSET : One-dimensional configurations?

17 DR. ROSZTOCZY: That's correct.

18 The second thing that we know is that looking at the
19 licensing code in RELAP 4, and the codes used by the vendors,
20 it becomes obvious that some of the basic physics needed to
21 predict or handle this phenomena is simply not in the codes.
22 So we know that the codes which are today being used for
23 licensing just don't have the capability to handle this
24 phenomena.

25 Therefore, going back to the available licensing

1 calculations cannot answer the question. We need some other
2 means to answer the question of whether it is important for
3 PWRs.

4 So we did the best that we could. We went to our most
5 elaborate and most difficult code, which is TRAK. We have
6 done calculations with TRAK in 3-D vessels for an actual PWR.
7 This is being done at LASL by the TRAK people. We had a pre-
8 sentation from them. We know the results they are getting.

9 The evaluation of this is that they do see this
10 phenomena actually occurring in their PWR calculations.

11 The fourth observation here is kind of a side obser-
12 vation, that using the RELAP 4 code, it did not predict this
13 behavior. There was a problem with the code, and in order to
14 resolve this, they have repeated the calculations, putting less
15 ECCS water into the system than actually was done in the past.
16 The purpose was to avoid the stacking up of water in some of
17 the boilers.

18 When they did this, then they got results which are
19 much closer to the test result. This is a possible indication
20 that maybe in the test some subcooled water is being by-passed
21 that doesn't enter the downcomer and doesn't enter into the
22 process the same way.

23 Summing all of this up, our present point is that,
24 yes, we are well aware of the fact that the heat flux in
25 Semiscale is larger than expected heat transfer in a PWR.

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Nevertheless, the phenomena being a real phenomena, we would like to know under what condition could this happen; and then we would want to compare that to the predicted conditions in a PWR and see if there is any overlapping.

And we know from calculations that if you change some of the other parameters -- not the heat flux but some of the other parameters -- you can step up the same phenomena. Basically if you get saturation in the downcomer, you get the phenomena.

For example, with a much lower heat flux than what was in Semiscale, like taking the heat flux of the PWR, but decreasing the subcooling of the ECCS water, the same phenomena are seen.

We also learned that the important test factors are that cold water enters in the actual systems in the hot legs. There are relatively small steam flows in the hot legs. In the cold leg, when you enter the ECCS water, there are relatively large steam flows.

This preheated water -- the water going into the downcomer -- is not on the temperature of the ECCS water but is already radiated. The heat flux is very important.

It's very important how much steam is generated in the reactor core. If in the reactor core the flooding stops momentarily because there is voiding in the downcomer and there is a possible flow reversal, then the steam generation is

1 going to change in the core unless you properly account for the
2 steam generation. You will get the correct pressure. If you
3 don't have the correct pressure, you don't have the correct
4 temperature. Some of these are put in the calculations by a
5 simple assumption. You have to put all the basic physics in
6 there and then see what happens. Once we have a tool in our
7 hand, then we can take into account the low influx of the PWR;
8 we can take into account the proper temperature, the proper core
9 behavior, the proper core lengths. The core length seems to make
10 a significant difference here. That may be why it was not as
11 pronounced in the earlier tests.

12 When you take all of this into account, you will
13 know under what condition it exists in PWRs, whether it's real
14 in terms of whether you expect to see it in an actual loss-of-
15 coolant accident. We think it's very important to resolve this
16 as soon as possible.

17 One other item that comes up here is that we have
18 provided a presentation last September to each of the
19 licensing boards who are faced with PWRs.

20 We have updated this information that I just pre-
21 sented to you. We have updated this. That was in May 1979.
22 There have been no licensing actions since May 30 because of
23 other complications, mainly Three Mile Island.

24 As we go back to our normal licensing work, this will
25 be introduced. I think the urgency is on to try to resolve it

1 as soon as possible.

2 Yes, it is possible that the outcome will be that it
3 is not important for PWRs. It is also possible that the out-
4 come will be that under certain circumstances it is important
5 for PWRs.

6 DR. PLESSET : Dr. Tong mentioned a very, very
7 important point, the so-called downcomer, the fact it wasn't
8 insulated. It's also one-dimensional downcomer; right?

9 DR. ROSZTOCZY: Yes.

10 DR. PLESSET : I know you are very much interested
11 in the multidimensional behavior. I know what your attitude
12 was about the German upper-plenum test facility. You were
13 very strong in having a 360 degree rather than using a plane
14 of symmetry with 180 degrees. You are quite sensitive to
15 multidimensionality of a downcomer?

16 DR. ROSZTOCZY: Certainly.

17 DR. PLESSET : Good. On that basis, I would say there
18 is another reason to be suspicious of the significance of this
19 Semiscale on 7-6 because of the -- well, it wasn't insulated.
20 Maybe that will take care of it; but even beyond that, the
21 one-dimensional nature of that injection system, what do you
22 think of that?

23 DR. ROSZTOCZY: That's exactly the reason why we
24 went to the three-dimensional calculations and went to the
25 actual size and actual geometry. We don't have the means or

1 capability to check out the large three-dimensional test.

2 Up to now, those three-dimensional calculations show
3 that this actually occurs; but again, please keep in mind,
4 because it is in a large system, because it is in three dimen-
5 sions, because the heat fluxes are different, what we see there
6 is not as pronounced and not as strong as it was in the test.

7 Now, the great value and the reason why we think
8 S-07-6 is a great test is because it's somewhat simpler than an
9 actual three-dimensional case and provides a good basis to
10 check the codes, whether some of the important physics is
11 present in the code and works correctly.

12 Once we check those out, then we can go and do the
13 necessary calculations either to show that we don't have to
14 worry about this problem or to show under what conditions do we
15 have to be careful?

16 DR. PLESSET : You discount the measurements and
17 calculations for LOFT?

18 DR. ROSZTOCZY: No. Nowhere is there any discounting.

19 DR. PLESSET : Because the effects aren't very
20 serious there as I guess from what we have heard.

21 I also gather you discount the calculations we heard
22 yesterday about Zion. They are rather detailed calculations.

23 Yes, Stan?

24 DR. FABIC: S-07-6 provides an interesting test
25 data, but I am very dubious that it is a good test to verify

66 1 the codes primarily because the heat flux from the walls of the
2 downcomer deployed in that pipe downcomer is really not known
3 at all. They measure temperature history, but that is not in
4 flux. You can specify anything you like and get the results.
5 That doesn't change the code.

6 DR. PLESSET : I think that's a third point, really.
7 I mentioned LOFT; I mentioned the Zion calculations. I think
8 this is a very difficult thing for a code which otherwise might
9 be quite useful. I think I agree with you on that point. But
10 this is a hard calculation to try to carry through with any
11 kind of confidence. If I understand you correctly, you agree
12 with it?

13 DR. FABIC: The boundary conditions are not known.

14 DR. PLESSET : That makes it very difficult.

15 We will come back to you, Zoltan, so you have the last
16 word.

17 DR. CATTON: I would like to add a comment on the
18 codes. If it's an instability mechanism, there are two solu-
19 tions. In the written code, you get one of them, not the other.
20 I don't know how you are going to carry out checking out the
21 codes for this particular problem.

22 DR. PLESSET : I think Theo has a comment.

23 MR. THEOFANOUS: Along the lines that Milton and
24 Stan were talking about, also I am puzzled, because I see your
25 Items 3 and 4. I would think that one-dimensional codes would

1 have this effect more pronounced than two-dimensional codes or
2 multidimensional codes. Yet, I think I got it from what you
3 presented that the TRAK 1, 1-D, was not able to predict the
4 phenomena. RELAP 4 was not able to predict that. You
5 emphasized that strongly.

6 Yet, you are saying you somehow were able to fix
7 the infinity track and are able to predict some resemblance at
8 least to the phenomena. Do you have any comments about that?

9 DR. PLESSET : Yes, maybe if you calculate long
10 enough --

11 MR. THEOFANOUS: If you try hard enough to do some-
12 thing, you get some resemblance.

13 MR. MATHIS: You get the answer you want.

14 (Laughter.)

15 DR. ROSZTOCZY: Okay. Let me start with Stan. I
16 didn't mean to imply that the only thing we need is the S-07-6
17 data as it stands and then you can qualify the codes against
18 that. No. Obviously, we need additional tests. Already some
19 of them have been done which have spoken to the downcomer.
20 New insulation is put into the downcomer to reduce the heat
21 flux to a low level so the influence of the heat flux will be
22 smaller than at the present time, and also you would get much
23 better cut-off.

24 But the fact that the system is one dimensional doesn't
25 mean that we cannot test the code against the system -- all the

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1 codes for licensing. They cannot predict with a one-dimensional
2 system. Then something is wrong with the code.

3 DR. PLESSET : That was Stan's point.

4 DR. ROSZTOCZY: It's a necessary but not sufficient
5 test.

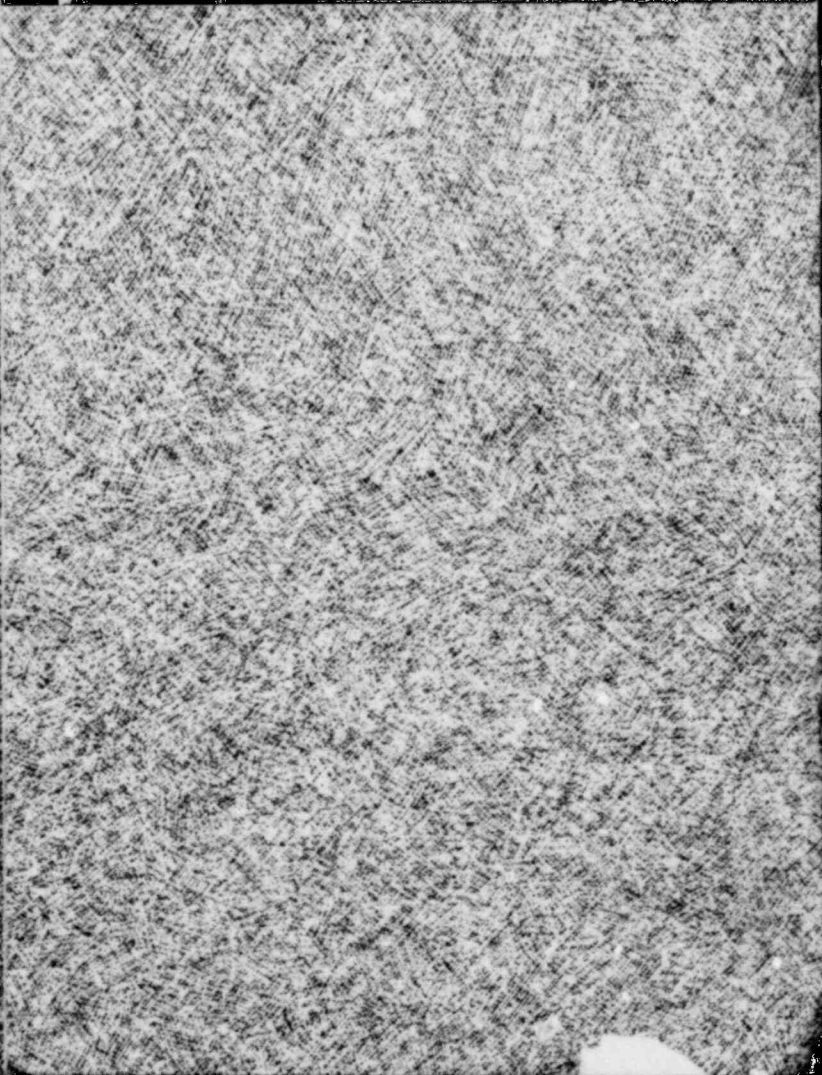
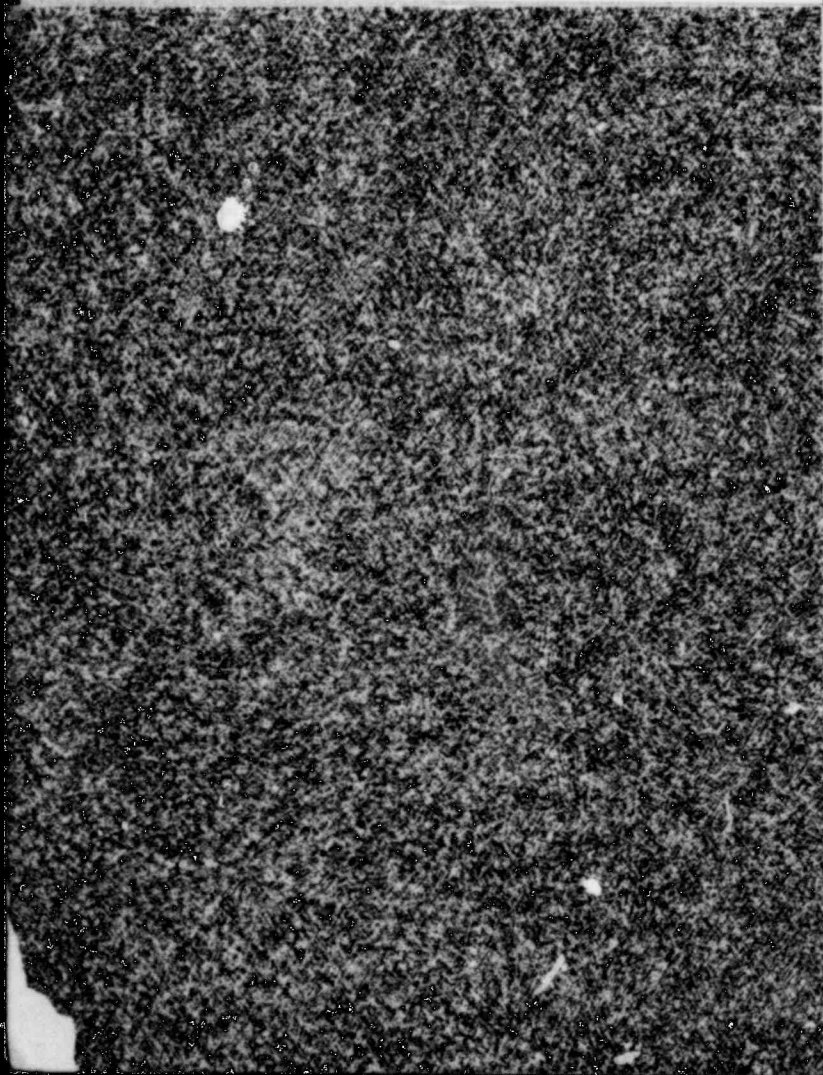
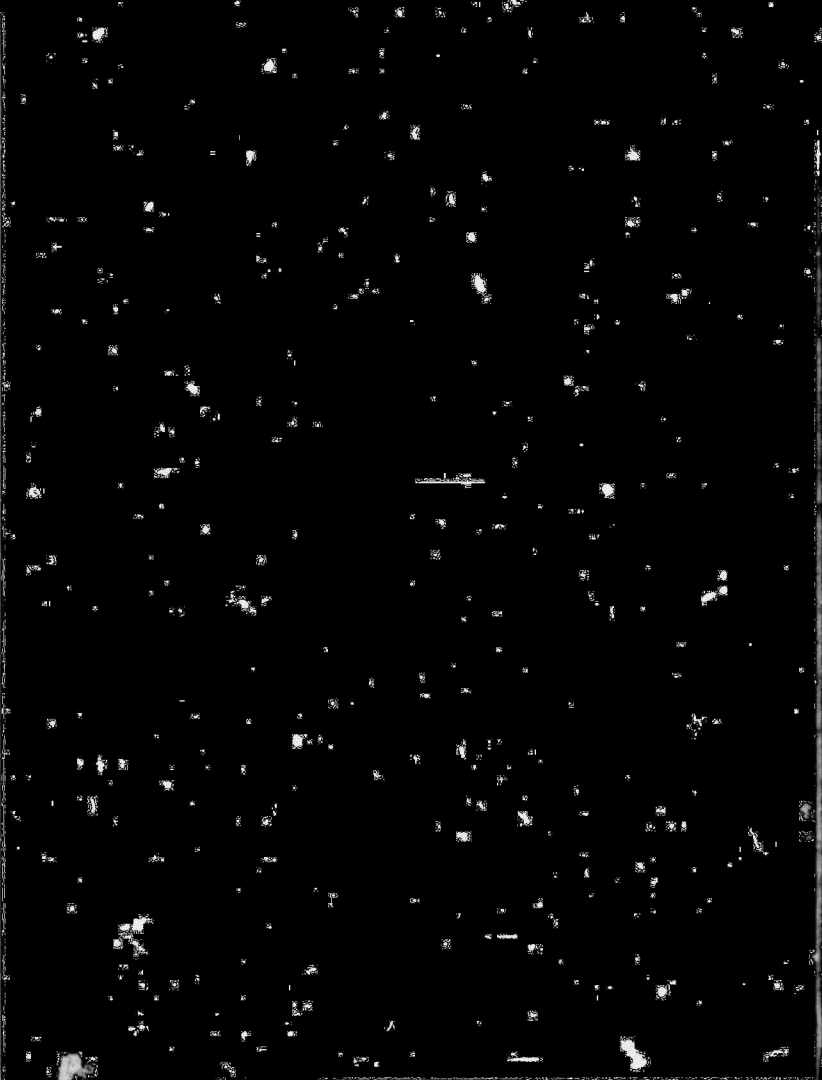
6 As soon as the data has been corrected and new test
7 results are available which don't have to be answered in terms
8 of the insulation, then you can check the codes against them.

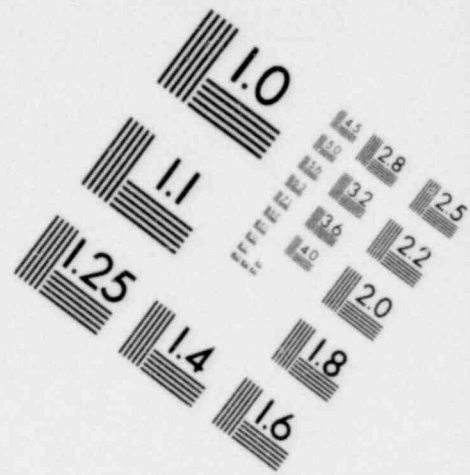
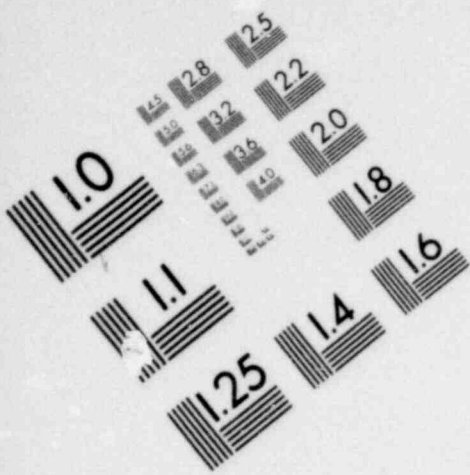
9 Let me go to Dr. Catton's comment as to why the codes
10 can't do it. There is a simple reason why they can't. Nobody
11 kept this in mind when they developed the code. We have spent
12 approximately or somewhat over \$100 million, we and the
13 industry, on code development for loss-of-coolant accidents.
14 Among the code developers, nobody had this in mind until we saw
15 the S-07-6 results. In each of the licensing codes, there are
16 basic assumptions which completely make it impossible for the
17 codes to predict this phenomena.

18 For example, when the steam generation, the steam
19 flow out on the top of the core is arbitrarily assumed to be a
20 fraction of the inflow into the core, then that code cannot
21 handle flow reversal, which brings saturated water into the
22 downcomer.

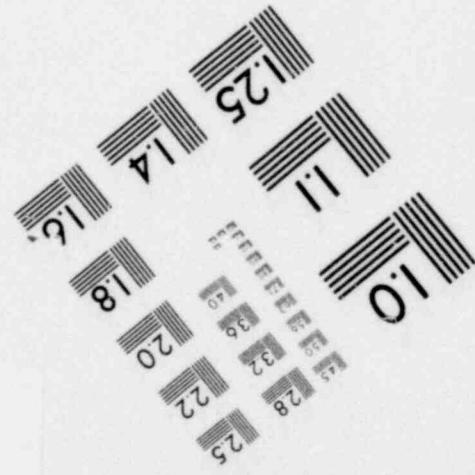
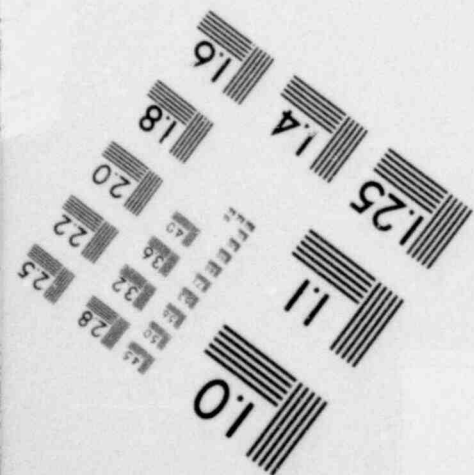
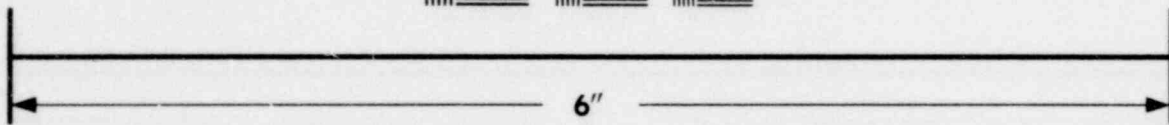
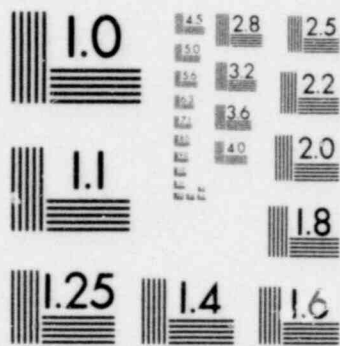
23 MR. THEOFANOUS: You are going farther. Why do you
24 limit your remarks to licensing boards?

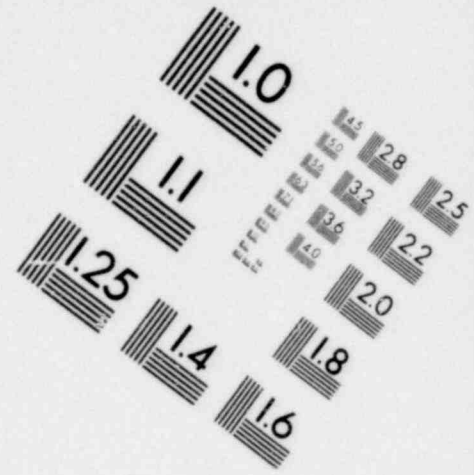
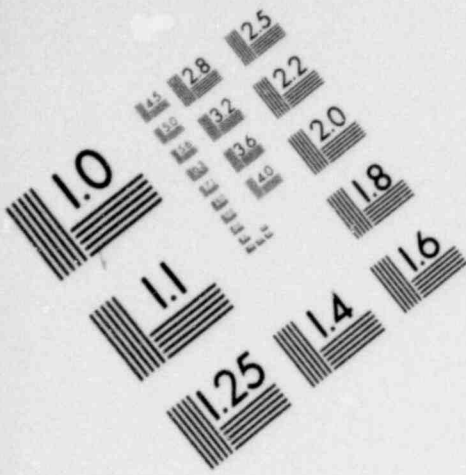
25 DR. ROSZTOCZY: I am going to address all of them.



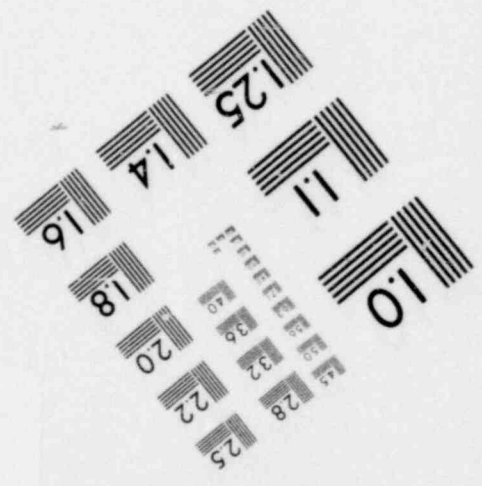
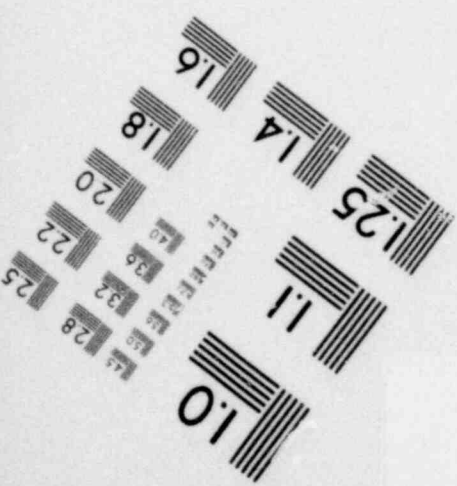
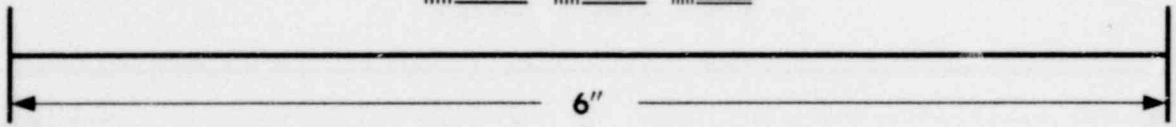
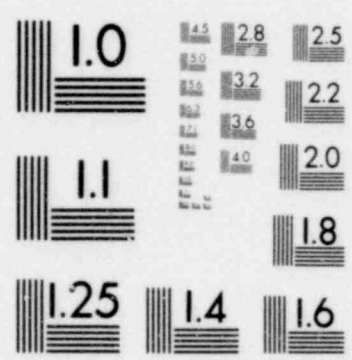


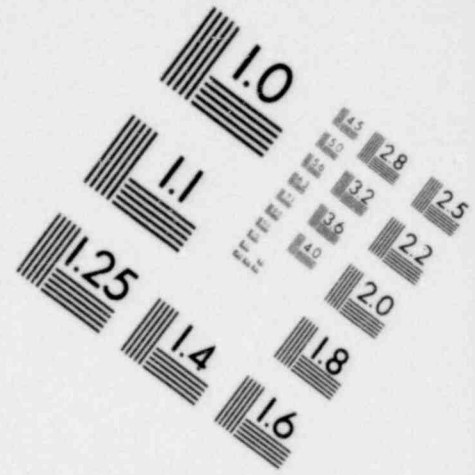
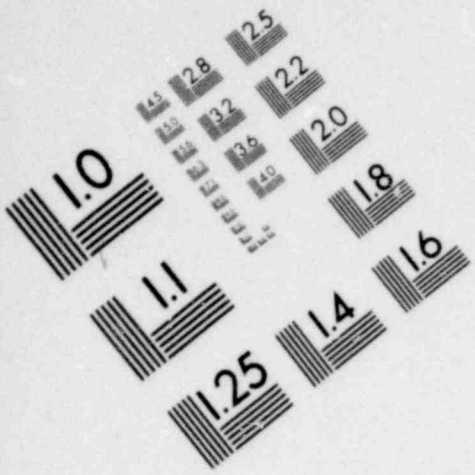
**IMAGE EVALUATION
TEST TARGET (MT-3)**



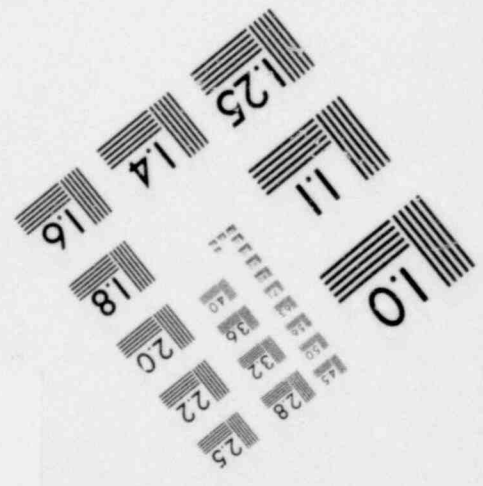
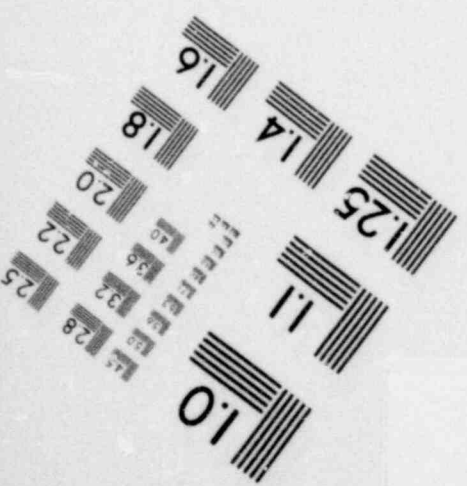
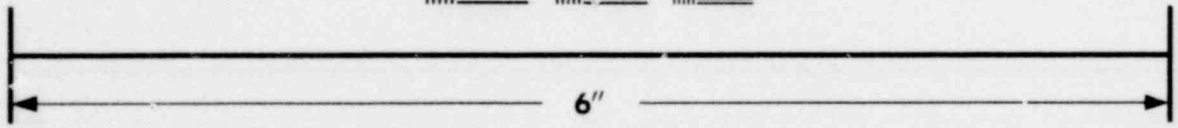
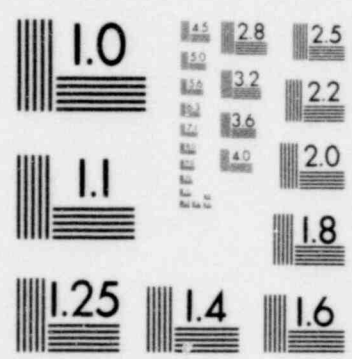


**IMAGE EVALUATION
TEST TARGET (MT-3)**





**IMAGE EVALUATION
TEST TARGET (MT-3)**



1 MR. THEOFANOUS: You still --

2 DR. ROSZTOCZY: That's the rather common assumptions.
3 Some of the licensing codes use that. They don't introduce the
4 cold water into the coolant. Therefore, they don't mix it with
5 the hot steam that's going around in the core. Instead, they put
6 the cold water into the lower plenum.

7 DR. PLESSET : It's a good place to have it.

8 DR. ROSZTOCZY: Part of the phenomena -- it would be
9 a good place in the real world if it would go there. It
10 doesn't go there, however, in the calculation that is being
11 put in.

12 There are some of these assumptions going into the
13 calculations which made the calculation smoother. They could
14 progress maybe on a shorter calculation time; but those
15 phenomena -- when we go to the more elaborate codes -- and I
16 think that was Dr. Theofanous' question -- the TRAK 1-D is
17 entirely different than the TRAK 3-D. The difference is not just
18 going from one dimension to three dimensions. TRAK 1 was using
19 a reflux model while TRAK 3-D handles the problem entirely
20 different.

21 The conclusion I got from the Los Alamos people working
22 with TRAK 1-D is that they are unhappy with the reflux model
23 in TRAK 1-D. They gave up on the prediction of this problem
24 until TRAK 1-D will be converted into a two-fluid mode next
25 year. Then they will come back and calculate it.

1 The basic problem is the two-phased flow model in
2 that version of the code. The two-phased flow model in TRAK
3 3-D is entirely different. They have more faith in that than
4 they have in the TRAK 1-D.

5 Right now they believe that the calculation that
6 they are doing for a PWR with the 3-D version and the code
7 two-phased model is a more appropriate model.

8 DR. PLESSET : Stan?

9 DR. FABIC: Just a point of clarification about
10 nomenclature.

11 As you are aware, the two-fluid model was used in
12 TRAK for anything inside the vessel. The reflux 1-D was for all
13 the loops outside the vessel.

14 They decided that when they looked at this problem,
15 to look at that pipe, which is called the downcomer, it was not
16 part of the vessel, as part of the loop. So they used the 1-D
17 flux model to describe the phenomena in the so-called down-
18 comer.

19 We have known that our reflux models, physics in the
20 reflux were not very good.

21 That's the reason why they were not happy with the
22 result of the calculation. Now they are trying to do it all
23 again.

24 That's the nomenclature.

25 DR. ROSZTOCZY: Yes.

1 DR. FABIC: There is no such thing as a TRAK 1-D.

2 DR. ROSZTOCZY: The main difference, as far as this
3 program is concerned, is that this is handled differently in the
4 calculation.

5 DR. PLESSET : I think Stan made a good point.

6 DR. ROSZTOCZY: The other concern we have is that
7 when the reactor designers designed the ECCS system, they did
8 not have this phenomena -- they didn't think of this phenomena.
9 It wasn't in their design calculations. Should this arise in
10 some of the loss-of-coolant accident, it could set some
11 requirement for the ECCS system. It's possible that those
12 requirements can be very easily met. The subcooling is a very
13 important parameter. It is important that you not put in
14 water than a certain temperature. I am certain you can cut it
15 out from the PWRs. If you know the subcooling is not hot
16 enough, it will not happen in a PWR. This is not a design
17 requirement. There is no such limitation on those plants,
18 depending on where the plant is located or what time of the
19 year they are at. There is different temperature water
20 available for ECC.

21 It is possible that what is available is sufficiently
22 low.

23 DR. PLESSET : I hope you don't make this requirement
24 yet to control the water temperature.

25 DR. ROSZTOCZY: We made one requirement. That one is

1 spelled out here. We have required the vendors to provide an
2 evaluation where they can show, (a) that the calculation is
3 proper even if it doesn't follow the actual physics, but that
4 that calculation is still appropriately conservative. Or if
5 they cannot show that, modify the calculations to account
6 for this phenomena should this phenomena come into play after
7 some break size or some calculation, that the calculation
8 would hold. That's the only requirement.

9 DR. PLESSET : Maybe we should go on.

10 DR. ROSZTOCZY: So. The only other point is that
11 we do need this analytical tool. We are encouraging every-
12 thing in that direction to see what can be done with that.

13 (Slide.)

14 The last group of data needs that I am going to talk
15 about relates to transients and accidents.

16 Transients -- and I mean here transients like loss-of-
17 feedwater transient, excess load type of transient, rod with-
18 drawal, turbine trip, station blackout, and so on, typical
19 transients recognized. These transients are a lot more mild
20 and the consequences are a lot more favorable than the con-
21 sequences of what we call accidents.

22 Because of this, we have been doing significantly less
23 in terms of core development, in terms of actual calculations
24 for loss-of-coolant accidents.

25 One recent test program performed on the Peach Bottom,

1 reactor -- this was in the spring of 1977 -- showed that the
2 calculational techniques for these transients, the error can
3 be a significant error.

4 Boiling water reactors, there is void in the core
5 due to normal operation. Once you go through on a transient,
6 then this void might collapse or might increase in value
7 which affects the reactivity of the core.

8 The problem found in the Peach Bottom test was that
9 the void collapsed; it wasn't correctly treated in the code.
10 Because of this, the code results gave us quite different
11 consequences than was measured.

12 At that time we issued a Board notification concern
13 on this, and this problem is still being resolved. General
14 Electric has developed a new computer code that accounts for
15 the phenomena, and they have shown that that code can predict
16 the test results obtained in Peach Bottom. They are going to
17 do new licensing calculations for those transients which have
18 void collapses.

19 Challenging test results for PWRs are not available.
20 As part of the start-up of the plants, there are usually some
21 transient tests. Those tests are oriented strictly to
22 activity. They demonstrate through those tests that various
23 equipment are working. They don't have the instrumentation for
24 the severity in the tests which provide challenging tests
25 for computer codes. So we do not have any real good data on

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1 PWRs. Because BWRs don't have a void in the core during
2 normal operation, we think they are probably less sensitive
3 and we think the accuracy of those calculations is significantly
4 better. Nevertheless, we would like to see some confirmation
5 of this.

6 We are seriously considering requiring some selective
7 tests on selective plants, both pressurized water and boiling
8 water reactors, to provide a data base for the verification of
9 the dimension of these codes.

10 Once we go to the accidents, we are reviewing the
11 steamline break and feedline break analysis methods. In this
12 review, we are again finding kind of a shortage of available
13 experimental data. There is some separate-effect tests
14 available to justify certain problems in the codes. There is
15 no good test to test the code in any sense.

16 Analysis methods for steamline tube rupture have not
17 been submitted and have not yet been reviewed in detail due to
18 some of the failures in operating plants. We are paying more
19 attention to this and are requiring the vendors to submit
20 calculational methods in the future. We are going to look at
21 those carefully.

22 Experimental verification will be needed. Our
23 feeling is that the best way to attack this is to have some
24 data on experiments to be done on an operating plant, and to
25 also provide some additional data on test equipment like LOFT

1 and the combination of the two should be sufficient to provide
2 that.

3 For the accidents, the steam generator dynamics have
4 been carried out. They are important.

5 Codes have been developed for that and should be
6 really representative.

7 These are basically the test programs that we are
8 looking for.

9 In closing, I would like to make just a few remarks.
10 What we have observed during the past couple of years is that
11 most of the information relative to plant safety evaluation is
12 coming from the experimental programs. Quenching of the core
13 is a good example of that; the voiding of the Semiscale down-
14 comer, the observed high steam generator in the EPRI test, and
15 the void collapse as it was observed in the Peach Bottom
16 test.

17 Unfortunately, the calculational techniques, including
18 our sophisticated calculational techniques, at the present
19 time are not good enough to predict this phenomena. So any
20 phenomena we are running into we might not have accounted for
21 is simply coming from the experimental.

22 At this stage of development we find that -- it is our
23 opinion that it is very important to keep on doing some of
24 this testing. It is very important to have test facilities
25 available should the need arise for some urgent testing. So

1 if some problem comes up somewhere, because we cannot test the
2 calculations to the DBS, it is important for us to have some
3 test facility available to check this out. We are hoping for
4 the continuation of Semiscale testing in the future.

5 I would like to point out one weakness that we have
6 observed during the test itself in the past few years. This
7 is the analytical support that has been provided for the test
8 programs. We fear that the analytical support hasn't been as
9 strong as it should be.

10 For example, the tests that are being done at GE
11 and San Jose where EG&G has the responsibility to perform that
12 analysis. For example, the computer code used for the
13 analysis of the test doesn't qualify to do some of the work
14 that the test was set out to do. We need a better code for
15 some of the tests.

16 The Peach Bottom tests, which were very significant
17 tests for boiling water reactor transients, there were no pretest
18 predictions done by any means available to NRC. The only pre-
19 dictions were done by GE. NRC as a whole did not make an
20 effort to do the pretest predictions. These tests are rather
21 expensive. We will see only a few tests of this sort. I think
22 it is important that we try to get them.

23 So I would like to urge both the contractors like
24 EG&G as well as the NRC to pay increased attention to the
25 analytical support work of this test, the data, the information

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1 that comes out that can be interpreted faster and more
2 accurately into the licensing procedure.

3 As you know, we have instructions from the
4 Commissioners to report to the licensing board every bit of
5 new information that we run into, even if we haven't had a
6 chance to resolve that, whether it is important or might not
7 be important for licensing process. If we operate in this
8 way, I think it's rather important to follow up on the experi-
9 mental points as to whether we know if they are important or not
10 important. That way we can keep the licensing process going
11 in an efficient manner.

12 That completes my presentation.

13 DR. PLESSET : Thank you, Zoltan.

14 With regard to reporting to the licensing boards, do
15 you report to the licensing board new data on heat transfer?
16 Mr. Nelson showed us this picture, the data bank. There are
17 some new things. Do you report those?

18 DR. ROSZTOCZY: We report to the licensing boards on
19 every information that could significantly -- that has a poten-
20 tial to affect the licensing of the particular plant.

21 DR. PLESSET : If we had another bit of information on
22 heat transfer like -- well, this Iloege was mentioned yesterday.
23 That could be very significant.

24 DR. ROSZTOCZY: Yes.

25 DR. PLESSET : Do you report that?

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1 DR. ROSZTOCZY: Whenever we run into something that
2 has a significant effect, we do. What was observed there is
3 that there was more heat transfer than possible in the calcula-
4 tion. That has been reported both in terms of the increased
5 steam generation and in terms of the possible improved heat
6 transfer. That has been reported in that sense to the
7 licensing boards, yes.

8 Now, if we run into a case which shows us that a
9 given licensing type of calculation is most probably better off
10 because of some new item that we found, but we have already
11 found the plant fully acceptable as it was, then the fact that
12 it's even better off than how we found it is not that important
13 in the sense, because we are 100 percent recommending to the
14 licensing board that the plant stand fully behind it. There is
15 less emphasis on reporting those things which make an already
16 acceptable case even better.

17 There is a lot more emphasis to reporting those
18 items which could possibly lead to other requirements beyond
19 those presently in the technical classification of the plant.

20 DR. PLESSET : So you don't put particular emphasis
21 in a report to, for instance, the LOFT results?

22 DR. ROSZTOCZY: The LOFT results are showing
23 quenching of the core earlier than it was expected.

24 DR. PLESSET : They also showed very low peak clad
25 temperatures altogether.

cs7° 1 DR. ROSZTOCZY: No. I am afraid we are not arriving
2 at that conclusion. As a matter of fact, we are looking at now
3 in some detail -- as you know, we have done a so-called
4 licensing calculation for LOFT tests; so did Exxon Nuclear
5 Corporation. When we take up from that calculation some of the
6 items which were introduced and it's specific to LOFT, but it
7 didn't exist -- namely, that the LOFT fuel is a low-density
8 fuel that goes through on much more fuel densification -- when
9 we take some of those atypicalities off, then the remaining
10 margin between our licensing process and the actual temperature
11 measured in LOFT is significantly lower than the margin intro-
12 duced into the ECCS hearing back in 1972 during the hearing.

13 We do have a concern that possibly the margin might
14 be lower than we all thought it throughout the years.

15 DR. PLESSET : Some LOFT might not be licensable?

16 DR. ROSZTOCZY: No. That is not so. We have learned
17 a lot; a lot of tests have been run. Lots of calculations have
18 been run since 1972. Seven years have passed. During the
19 seven years -- these were very intensive years in terms of
20 loss-of-coolant accidents. A lot has been learned. Maybe
21 this is an appropriate time now to pull all this information
22 together, sit back, think about it a little bit, see what have
23 we learned, and see if our licensing approach that was passed
24 on back in '73 needs any update.

25 DR. PLESSET : Well, I must admit to being a little

80 1 taken aback by what you are saying and the kinds of things
2 that should be of concern for the licensing boards by way of
3 helping them, which is what one is supposed to do; right?

4 DR. ROSZTOCZY: We have two responsibilities. One of
5 them is to help them in any area wherever they ask for our
6 help. The second one is simply to inform them. It is our
7 responsibility to bring to their attention various things that
8 are going on in various results, calculational results, that
9 we have obtained which might have an effect on the licensing
10 that they are faced with. It is our responsibility to provide
11 this for them.

12 We also provide it with an appropriate explanation
13 and an appropriate -- presenting the problem in an appropriate
14 manner.

15 DR. PLESSET : One thing I am trying to maybe get at
16 is that I think Semiscale certainly is a useful and significant
17 test facility; but I would regard it more in the class of a
18 separate-effects facility rather than as really giving
19 information with not a lot of qualification and reservation in
20 its applicability to a PWR. This is what I was trying to get
21 at. I don't know if anybody else agrees with me or not. If
22 so, maybe they can -- or disagrees. Please feel free.

23 Does anybody else want to comment on this last point?

24 MR. ALLEMAN: I don't quite understand how what you
25 said, Zoltan -- although the predictions for LOFT look to be

91 1 conservative, you don't think they would be conservative for
2 licensed reactors?

3 DR. ROSZTOCZY: Let me say maybe a few more words
4 about that so we understand it better.

5 Back in 1972 and '73, when the hearings were going
6 on, and there was various information in the hearings, calcu-
7 lations were introduced into the hearing by most of the parties
8 where they showed what kind of peak clad temperature they
9 calculated as a so-called model. They showed what kind of
10 peak clad temperature they would expect to see in the real
11 world.

12 There have been some calculations of this sort. I
13 believe Dr. Zyrmak has done some. There was information avail-
14 able. When we went to the LOFT test, one way one can get more
15 information of this sort is to do a calculation with the
16 licensing codes for LOFT and compare the results of that calcu-
17 lation to the test. It has been done. It was done by running
18 the test. The calculation was performed.

19 What has been done post-test and is still going on
20 is to take this calculation and then take off some of the
21 assumptions one by one to try to find out how much is accounted
22 for by the calculation.

23 Two licensing calculations were done by LOFT, one by
24 Exxon, one by the NRC. The Exxon calculation -- if you want,
25 I can pull out the slide.

1 DR. PLESSET : That's all right.

2 DR. ROSZTOCZY: The accident calculation was higher
3 than the actual measured temperature; but the difference be-
4 tween the two was not as large as the difference shown in the
5 hearing calculations or in the Zyrnak study that was done.

6 The difference between the two calculations was large,
7 was as large or larger than what has been shown in 1972 and what
8 has been shown in Dr. Zyrnak's work; but there were a few
9 things in the calculation which are very unique to LOFT.

10 One of the problems is that there is a large penalty
11 on fuel densification. Because the LOFT fuel has been manu-
12 factured a long time ago, it's a low density fuel. If somebody
13 would introduce that type of fuel into a reactor, we would
14 still require the same penalty on fuel densification. So they
15 put into the LOFT calculation this fuel densification penalty.

16 If we remove that and replace it as much as we can
17 with something which would be more appropriate, our calculation
18 comes very close to the actual calculation.

19 So the extra margin we have there is due to -- I
20 think there was one other factor. It appears there is no major
21 disagreement with the Exxon calculation now; but the margin that
22 you see there is not as large as we have seen before.

23 Again let me emphasize what was emphasized earlier:
24 One has to be careful. One can't pick up this number and
25 ultimately carry it to a PWR. You have to check and look at

1 the fact that there are other things which might provide a
2 margin on the PWR but does not show up on this test.

3 After you compare all of those, then you can say my
4 evaluation today is that all these assumptions provide margins.

5 MP. ALLEMAN: You say the margin is statutory?

6 DR. ROSZTOCZY: The basic principle which has been
7 stated by the Commissioners in the comments that they made at
8 the time when they issued the ECCS criteria was that there
9 should be sufficient margin to cover the uncertainties of the
10 calculations, plus a safety margin beyond that. It has never
11 been specified in terms of how big the extra margin should be.
12 Instead, a judgment was made that the requirements specified in
13 Appendix C should be put into the calculation, all of them
14 provide sufficient margin to meet these principles.

15 What I am suggesting is that maybe are getting a lot
16 more information available today. Maybe we are getting to a
17 phase where one can specify this a little bit more accurately.
18 Maybe one can get a handle on what is the uncertainty of our
19 codes.

20 Once you can quantify both of these, you can compare
21 one against the other and see what comes out.

22 My personal judgment would be that nothing would come
23 out of it.

24 Then we would change our way and how to handle this.

25 For example, one might find that we have enough

1 margin in the large-break calculation but not in the small-
2 break calculation. Maybe you change the way you handle the
3 small break.

4 MR. ALLEMAN: Our uncertainty of the code is less
5 now than it was?

6 DR. ROSZTOCZY: No. Unfortunately, I cannot say that.
7 Again based upon the limited knowledge available, people tried
8 to formulate some idea of the uncertainties. I don't think
9 those are accurate. There are possibly some statements about
10 that in the hearing notes.

11 This margin that was shown, as I recollected, was on
12 the order of 500, 600 different margins. So that was supposed
13 to cover the code uncertainties.

14 Now, if we go back and turn to the people working
15 with the code and ask how good these codes are, then I think
16 they are good to something like plus or minus 500 degrees.

17 That's not very good as a result at this time.

18 We learned a lot about the codes, but the overall
19 picture of the codes is not that up to date.

20 Maybe it is because of our ignorance.

21 DR. PLESSET : You indicate a lot of reservations,
22 which are commendable, regarding the optimism with which we
23 should apply LOFT data to a PWR. I wish you had the same amount
24 of negative reservations regarding the applicability of
25 Semiscale to a large-scale PWR.

1 (Laughter.)

2 DR. ROSZTOCZY: Dr. Plesset , I do; I do. You ex-
3 pressed most of those before I got to them. Our vision of
4 Semiscale is no one should take the 400 seconds delay. That
5 would be complicating the problem. The only thing that is
6 meaningful is to have an understanding of the processes that
7 the heat plays, then feed those into our extrapolation tech-
8 nique which we have as computer codes and then see with these
9 extrapolation techniques what is the prediction for a PWR.
10 We have stated that very carefully in both the Board notifica-
11 tion notice, which was issued last year in September, and also
12 in the update issued on May 30 of this year.

13 We are skeptical. We do not know if this phenomena
14 has a significant effect for PWRs. We would like to know it
15 and know it as soon as possible. For that reason, we find it's
16 a very urgent problem. We hope to resolve it fast. If we can
17 resolve it in two months, put it behind us, we are better off.

18 DR. PLESSET : I think we can go down the line.

19 Ivan?

20 DR. CATTON: I have a couple of comments. First, I
21 would like to take an opportunity to emphasize what I feel is
22 a need for more balance between experimental analysis. I think
23 you were bringing this out earlier, Zoltan. I have the same
24 sort of reaction.

25 Also, Semiscale is a highly 1-D system. It serves,

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1 as far as I can tell, as an excellent research tool; but with
2 what I feel is in the past a lack of this kind of balance,
3 many of the scaling questions are not answered. I just don't
4 see how tests like the S-07-6, without knowing how to scale
5 from one system to another, other inadequacies, can be viewed
6 without a great deal of qualification.

7 However, I do, I think, side with you on your view of
8 the 400 degrees. I think it's a serious question and probably
9 ought to be looked at, but not without stating all of the
10 qualifications.

11 DR. ROSZTOCZY: If you would develop a model from
12 basic physics, just to start physically with that, and you
13 compare it, you put it into one dimensional code, and compare
14 it against the Semiscale result, and get back the test result,
15 it would give you some comfort for using that code.

16 DR. CATTON: I feel somewhat unsettled that we have
17 a multimillion dollar code program, a 1-D facility. In a
18 sense, I am not surprised. It is indeed an instability. I
19 wouldn't expect the code to predict it. I think you have to
20 look for instabilities to predict.

21 DR. PLESSET : Theo?

22 MR. THEOFANOUS: In the interests of time, I don't
23 want to ask Zoltan a question, because he might take a very
24 long time to reply.

25 DR. PLESSET : So you are making a statement?

cs87 1 MR. THEOFANOUS: I want to make a comment.

2 One, I agree with you, Milton, on your view of
3 Semiscale; I think it is a useful facility. It is the kind of
4 thing that can trigger our imagination to see certain things
5 and think about the problem. However, as we learned many times
6 in the past, many times we have to go back and explain things
7 away that are not relevant to PWRs instead of the other way
8 around.

9 So every time you look at Semiscale results, I think
10 you should take it with a grain of salt.

11 Secondly, we should try to explain them. I think
12 that we should be very careful before we try directly to apply
13 them to PWRs.

14 As far as Zoltan, I want to disagree with you, Zoltan,
15 with your rather grim view of how much we learned with the new
16 codes and with the new analytical efforts that we have been
17 over for the past three years. I myself found them very useful
18 and relatively large contributions to the understandings of
19 loss-of-coolant accidents.

20 DR. PLESSET : Thank you; particularly the last
21 comment which I echo which should relieve the pressure building
22 up in Stan. He didn't even say anything.

23 (Laughter.)

24 DR. PLESSET : Thank you, Zoltan. You certainly
25 stimulated us. That's good.

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1 I think we have to look at the program a little
2 bit, and Mr. Mathis, the other ACRS member, has been a big help
3 to me.

4 What he is suggesting is perhaps we have 45 minutes
5 for program accomplishments, with emphasis on TMI work. As he
6 pointed out, we are pretty much up to date on these things.
7 I wondered if we might not omit that, perhaps get another
8 update at another meeting, if that will not be too painful.

9 We do want to hear the discussion of the scaling.
10 We are all very concerned with that. Perhaps while you mull
11 that over and accept that negative offer, we might take a break
12 at this time for 10 minutes.

13 (Recess.)

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1 DR. PLESSET: We will go to a presentation on
2 Semiscale scaling that Mr. Larson will present.

3 MR. LARSON: Gentlemen, my name is Larson. I am
4 going to have the honor today to discuss with you the
5 controversial topic of scaling.

6 (Laughter.)

7 DR. PLESSET: I am glad to hear that opinion.

8 MR. LARSON: One thing I would like to point out
9 before I start is that yesterday's comments necessitated the
10 addition of a few slides to this presentation. I think you
11 should have a supplemental copy.

12 (Slide.)

13 It is obvious that as long as I have worked for
14 Semiscale, there has never been a complete presentation on
15 the scaling of the system. any of the systems, the
16 philosophy, the approaches, the different criteria, and the
17 comparables. I think the interest that is obviously
18 apparent here dictates that I go through some of those types
19 of discussions.

20 It's also clear there is a considerable amount of
21 interest in scaling from a standpoint of extrapolation of
22 data from Semiscale and/or LOFT to the larger-scale
23 facility. I would like to make it clear at this point that
24 in the past we have not advocated extrapolation of data from
25 the facility. We don't necessarily intend to in the

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cs/dd 1 future. We realize by virtue of the scaling — and I think
2 there was a comment that Dr. Plessett made yesterday — by
3 virtue of scaling, the system is and has problems of one
4 nature or the other. We have to consider those things.

5 We also need to look at scaling from the
6 standpoint of how good is the data from a code standpoint,
7 usability of the codes, for development and assessment and
8 verification, whatever you want to call it.

9 To address some of these concerns, I would like
10 this morning to go over three areas:

11 Number one, the approach taken in Semiscale as far
12 as scaling is concerned. I would like to discuss three
13 different techniques that could have potentially been used
14 and then discuss the advantages, the disadvantages of those
15 techniques.

16 I'd also like to discuss the criteria that we have
17 established for scaling of the MOD 3 system and then discuss
18 a few of the compromises, resolutions to those compromises,
19 that we think we have some kind of a handle on at the
20 present time.

21 As far as the discussion of compromises, I will
22 discuss very briefly some of the large-break scaling. I
23 will call them distortions for lack of a better word.

24 I will also discuss some of the things we are
25 currently looking at for small breaks.

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1 It is not my intention here to answer all of the
2 scaling problems and questions. I am not at this time
3 prepared to do that. What I intend to do is give you an
4 idea of the kinds of analyses that we are doing and the
5 direction that we are taking at the present time.

6 (Slide.)

7 First of all, I think it is probably advantageous
8 to start with a definition of scaling under which we labor.
9 That is perhaps simplistically stated as the technique
10 whereby an important phenomena in some reference system are
11 simulated in a smaller scale, smaller geometry system.
12 Scaling is necessary for obvious reasons of cost, et
13 cetera.

14 I think, as I mentioned earlier, Dr. Plesset made
15 a comment yesterday that was appropriate, in that when
16 scaling there are certain things that you have to keep in
17 mind.

18 Number one, in general I don't think it is
19 possible to maintain geometric, dynamic similarity in a
20 scaled experiment for all kinds of concern. It is simply
21 not possible. An example of this is the difference between
22 blowdown and reflood. For blowdown, surface area-volume
23 ratios may not be as important as they are during reflood
24 when steam generation rates are extremely important.

25 Another concern or something that must be kept in

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1 in mind is that distortions will exist in the system where
2 multidimensional effects are present. That is a result
3 simply of the fact that some of the scaling considerations
4 that we work with dictate the need to maintain elevation and
5 perhaps also volume.

6 That introduces L/Ds in the system that are large,
7 hence causing the system to be largely one-dimensional.

8 (Slide.)

9 There are at least three possible scaling
10 techniques, probably more. I would like to discuss some of
11 those at the present time and then indicate why we picked
12 the particular technique that we did.

13 The first technique is linear scaling, whereby
14 L/Ds from the referenced system are maintained in the scaled
15 system. This technique has its advantages and
16 disadvantages, like the other three I will discuss, some of
17 which are in linear scaling. Acoustic times can be
18 maintained, given some time-scaling factor. The technique
19 has the advantage that it's been used in small-scale,
20 steady-state systems somewhat successfully.

21 It also has disadvantages from an integral
22 transient standpoint in that time scales are shifted so that
23 rate processes change, such as volume production, flashing;
24 extremely small linear dimensions result, also area and
25 volume. They change by a scale factor squared with the

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1 other. We also have the problem that, as an example, in the
2 linear scaled system of 2 megawatts, such as Semiscale, you
3 have heater rods that are extremely small, the size of a
4 piano wire. It's utterly ridiculous to scale from that
5 standpoint. You can't get 2 megawatts into a bundle that is
6 40,000 piano wires high.

7 Dimensionless numbers are a technique. This has
8 been used in mechanics for many years. It has a sound basis
9 from the pi-type theory or equations in motion. In a
10 transient facility, however, there are many numbers to
11 consider. The controlling numbers may not be the same
12 throughout the whole transient. I think, as an example,
13 Barnett tried to scale — I think successfully — CHF with
14 dimensionless numbers. He wound up with 14 dimensionless
15 numbers.

16 In a system such as Semiscale, there are
17 compromises, and it is not possible to maintain
18 dimensionless numbers such as Reynolds numbers for all
19 periods of time. Reynolds numbers are powerless for that
20 matter.

21 Volume scaling is also a viable technique and the
22 one in which we have used — it is also used to scale the
23 LOFT system. It has the advantage that the time scale
24 relative to the referenced plant is preserved. That's
25 important from rate process control, production quality,

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1 maintaining transitions from subcooled to saturated flow,
2 for example. That occurs at the right points in time.

3 We also have some considerations in that for
4 loss-of-coolant type transients. We need to have the proper
5 resistance distribution in the system so we get all splits
6 where we think they will occur in the larger chain.

7 Volume scaling allows you to do that by
8 maintaining volumes in the system and thereby producing
9 areas in piping, say, that are too fat, such that you can
10 work with them to provide the right resistance
11 distribution.

12 Volume scaling is not without its problems, of
13 course, just like all the other techniques. You can't
14 simultaneously maintain elevations, volumes, length-area
15 relationships, and resistance at the same time. So by
16 nature of the technique, there have to be compromises.

17 Another perhaps disadvantage of volume scaling is
18 acoustic transit times are not maintained simply because the
19 links are short relative to the reference point. That can
20 have an effect when subcooled decompression loads are
21 concerned.

22 (Slide.)

23 I would like now to discuss the MOD 3 scaling. As
24 I said, we picked the volume scaling approach. The
25 referenced plant, the design of the loops, was the

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1 Westinghouse Trojan plant. The design for the vessel was
2 basically taken in a scaled sense from a Westinghouse UHI
3 design. That was originally the purpose of the MOD 3
4 system, to look at UHI considerations.

5 I would like to point out at this time that the
6 initial scaling of the facility was done by EG&G personnel.
7 We had a lot of help from consultants, reviews with the
8 vendor, and also NRC personnel. The final design was
9 reviewed both by NRC and also our review group. So it's
10 been a joint effort. There have been a lot of compromises.
11 We are all aware of that fact.

12 The next slide simply presented the results of
13 volume scaling facility. I don't intend to belabor this
14 slide any.

15 . (Slide.) -

16 The important columns, I think, are the two listed
17 percentage totals. It is simply a comparison of the
18 relative distribution of volume in the system, in the
19 referenced plant — plants, I should say — and the MOD 3
20 system. It indicates essentially that we were quite
21 successful with the exception of the guide support tubes of
22 providing the right distribution of mass and volume
23 assistance.

24 As I stated before, we considered that important
25 because the distribution of mass at a particular temperature
indeed determined the flashing characteristics and

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1 indeed determines the flashing characteristics and
2 decompression characteristics of the system.

3 An obvious question that comes up is what kinds of
4 implications does volume scaling have? We, I guess I should
5 say, labored under what we called modified volume scaling.
6 That is, we did endeavor, as the previous slides showed, to
7 maintain volume distribution in the system. We also had the
8 additional requirement that we wanted to maintain elevations
9 of the system where possible.

10 (Slide.)

11 The intact loop, as I will discuss later, we had a
12 Type 1 steam generator which is short. Elevations relative
13 to the referenced plant are not maintained there. In any
14 event, the attempt was made to maintain geometric and
15 dynamic similarity in the primary heat transfer regions of
16 the system, number one, the core, and in the steam
17 generators.

18 The compromises that that, by virtue, produces are
19 that the piping line are not maintained; the piping area,
20 hence diameter, is not maintained. As I said earlier, with
21 the use of orifices, one can provide the right resistance
22 distribution in the system, and we considered that more
23 important in terms of providing the right flow splits, mass
24 flow distribution, than maintaining acoustic transit times.

25 Another particular distortion that volume scaling

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1 introduces is that piping surface area is not maintained. I
2 think we have all seen the results from S-07-6, the
3 distortion in the surface area to the volume, which I will
4 briefly touch on a bit later, what that does to the
5 characteristics in the downcomer in terms of ECC
6 penetration.

7 Another thing I mentioned earlier is the concept
8 of the L/Ds being large, hence causing the system to be
9 primarily one-dimensional. I say "primarily" because we
10 have run some tests where even in our small bundle, which is
11 about 6 inches in diameter or less, we have seen some
12 multidimensional cooling where we injected steam into the
13 upper plenum.

14 The question that you may ask right now is, "Well,
15 how well does volume scaling really work?"

16 (Slide.)

17 This slide shows a comparison of two counterpart
18 tests, one LOFT L2-2, which you saw the results from
19 yesterday, and results from Semiscale MOD 1 experiment.
20 This is not a MOD 2 experiment. S-07-2 experiments
21 were designed to be counterpart tests, to actually look at
22 the question of how well was the scaling accomplished.

23 I think this slide indicates that we indeed do get
24 similar break flow characteristics in the two facilities.
25 Now, the Semiscale flow rate here -- the flows have been put

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1 on a common basis by dividing one by the volume ratio of the
2 two. You can see that there is some difference here during
3 the subcooled blowdown. One must keep in mind the
4 uncertainties on both the Semiscale and the LOFT data. I
5 think a good estimate of that is plus or minus 20 percent.
6 So that the two do indeed overlay here.

7 The end result of that in terms of system
8 depressurization characteristics is shown on this slide.
9 Indeed, we do get depressurization characteristics as a
10 function of time that are similar in the two systems. We
11 have some confidence that we are getting the right rate
12 processes, energy generation rates, quality production, at
13 least for blowdown in the facility, so that there is some
14 basis for believing that volume scaling is a viable
15 technique.

16 There are slight differences here. There were
17 some very slight differences in initial conditions primarily
18 in hot-leg temperature between the LOFT and Semiscale
19 experiments. Our experiment we started with a 580 K
20 hot-leg temperature; in LOFT, the temperature was about 500
21 — excuse me, in our experiment we started with 587; LOFT
22 was 580. The saturation pressure of the system was a bit
23 different.

24 I would like now to briefly discuss a couple of
25 the MOD 3 components.

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1 that. The spacing in the core is a little bit off.

2 We do have structural surface area that is larger
3 than that desired from a scaling standpoint. As I said, we
4 did try to mitigate those consequences through the use of
5 insulation which we discovered now does not work to our
6 satisfaction. We are building new insulators to hopefully
7 combat that problem.

8 The differences in the rod properties are handled
9 through two techniques. The electrical power control is
10 controlled in a rather sophisticated fashion to hopefully
11 match the heat flux that we expect from a nuclear rod during
12 a transient where the nuclear rod is seeing the same
13 boundary conditions. There are two techniques for doing
14 that.

15 We are currently experimenting with an on-line
16 computerized power control which will determine the power as
17 the test is being conducted. There is another technique
18 whereby we iterate using results from one test, taking heat
19 transfer coefficients, and then off-line determining what
20 the power profile should be and using that in this test.

21 There are analytical considerations here in that
22 conduction calculations were done to help identify the need
23 for increased insulation and to mitigate the structural
24 surface area problem.

25 (Slide.)

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1 The intact and broken loops, as I mentioned
2 before, have the scaling criteria of hydraulic resistance,
3 volume distribution, and also pump suction leg depths. We
4 consider the pump suction leg depth important from a
5 structural standpoint.

6 I think you heard in the LOFT presentation
7 yesterday that that may not be quite as important as we
8 thought in the past. The only small-break experiment we
9 have ever done, S-02-6, indeed showed that that was a
10 consideration and did indeed determine some of the core
11 recovery characteristics; although in that particular
12 experiment core recovery was not anything substantial.

13 We did, as was mentioned earlier this morning by
14 Zoltan, run an additional small break. However, it is a
15 standard problem. We have not analyzed that data and
16 established the influence of the pump suction leg.

17 There are compromises in the loops in terms of, as
18 I said, lengths, flow areas, and again the surface area. We
19 also had instrumentation that influences such things,
20 perhaps, as the flow regimes and resistance. These
21 treatments have been both experimental and analytical, as
22 you will see in a moment.

23 We do hydraulic resistance tests, single-phase on
24 the system, to determine that the resistance is what we
25 desire. Conduction analysis has been done to look at the

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1 influence of the surface area relationship.

2 (Slide.)

3 One of the big concerns perhaps in a small break,
4 especially, is the steam generator characteristics. As I
5 think you are probably all aware, in the present MOD 3
6 system, we have two different type generators, one called
7 the Type 1, basically scaled for LOFT; and a Type 2,
8 basically scaled for a PWR. It contains full elevations and
9 pressure drops.

10 That in itself constitutes two compromises.: One,
11 length effects for the intact loop; number two, the broken
12 loop has an oversized secondary volume. That is a
13 consequence of two design considerations: One, the fact
14 that we design the generator with the thought in mind we
15 would like to use it in the intact loop as well as the
16 broken loop; number two, it is extremely difficult without
17 the use of filler pieces in that generator, just by virtue
18 of the small number of tubes and the height requirements to
19 get the secondary volume correct.

20 We are currently looking at the fixes for that. I
21 have some results that I will present later, I think, that
22 show we may be able in the near term, anyway, to drain the
23 generator carefully as a function of time, to provide the
24 right surface heat transfer area as a function of level.
25 The long-term fix for that is, of course, to attempt to put

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1 fillers in the secondary to maintain the right volume.

2 (Slide.)

3 I would like now to discuss briefly — and I don't
4 intend to dwell on this, simply because it's not a topic of
5 primary concern — the scaling considerations for large
6 breaks. I think there are necessarily positive points by
7 virtue of our scaling criteria. Some of these are listed.
8 I mentioned probably all of them earlier in the
9 presentation.

10 We do have the right resistance distribution,
11 which is important in terms of flow splits, particularly in
12 the core, whereas the stagnation point — this question came
13 up several times yestereay.

14 Our energy and mass distribution is correct, which
15 provides for the right rate processes, the right amount of
16 relative energy exchange.

17 Our core geometry in terms of pitch, rod diameter,
18 and length is correct. By virtue of control of break area
19 to volume, we can get the right time scale; and we have
20 attempted, where we can, to maintain elevations, relative
21 elevations.

22 Our vessel is full length, as is our downcomer,
23 which should provide proper elevations for such things as
24 reflood head and the like.

25 There are particular concerns. That includes

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cs/dd 1 metal-stored energy and surface area to fluid volume
2 ratios. The loops are fat, so the velocities are not
3 correct. There is a concern about pump characteristics and,
4 of course, steam generator characteristics.

5 I would like to address a couple of these things
6 now with some of the data we gathered from tests in the
7 past.

8 (Slide.)

9 I think most of you have probably seen this famous
10 slide before. It is a comparison of the heat flux or
11 estimation of heat flux from Test S-07-6, a
12 conduction-limited calculation for a PWR downcomer that was
13 assumed to see the same fluid temperature condition as was
14 experienced during this experiment, and was
15 conduction-limited.

16 Also, a calculation with our new insulator design,
17 the honeycomb, which, by the way, has a conductivity, an
18 overall conductivity of about .09. Just as a point of
19 interest, the honeycomb conductivity is on an order of 2.
20 It is not a very effective insulator. As a matter of fact,
21 it is a reasonable conductor.

22 I think the results from this calculation are
23 encouraging in that it shows what was about a factor of
24 three or more higher in heat flux in that experiment
25 relative to that estimated for the PWR. We are now down

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1 in the ballpark where the heat flux will be quite a lot
2 closer, at least the calculations so indicate.

3 I would like to point out at this point that there
4 are some scalings that have gone on to make these
5 comparable.

6 We still have a a surface area to volume area show
7 that is about a factor of 8 in the downcomer. What has been
8 done is that the Semiscale values of heat flux have been
9 multiplied by 8 to put them on a basis comparable with the
10 PWR.

11 The effects of pumps in a large-break blowdown, I
12 think, were discussed in some detail yesterday by John
13 Linebarger.

14 (Slide.)

15 We in the past have found similar things. This
16 slide shows a comparison of the downcomer flow. This is
17 from a RELAP calculation, in the case where the pump speed
18 was left on, and also in the case where the pump was
19 tripped. We have also done two experiments during the
20 course of testing in Semiscale Test Series 6 where we
21 experimentally investigated this and found a similar
22 phenomena.

23 The product of this was a change in the DNB
24 characteristics in the core, so that in the case where the
25 pump was tripped, you have a delayed DNB, by virtue of more

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1 flow into the core or out of the core.

2 Where the pump was left on, it appears that the
3 intact loop hot leg was provided enough flow to satisfy the
4 breaks so that we got a low enough flow in the core to
5 provide an early CHF.

6 That is the effect of pump trip versus no pump
7 trip.

8 There is also, I think, an analysis done in terms of small
9 pumps versus large pumps, Semiscale versus LOFT, and what
10 the effect of the degradation in the two pumps is. I think
11 for a large break in general, both pumps degrade
12 significantly by between 5 and 7 seconds; so those kinds of
13 considerations are not important in a large break. They
14 are, however, for a small break, as we will see in a
15 moment.

16 (Slide.)

17 We have also looked at the influence of steam
18 generator heat transfer during the course of MOD 1 testing.
19 This slide shows three results, one from a RELAP
20 calculation, two from experiments that were conducted back
21 in the isothermal test services. Test S-01-5 was run with
22 nitrogen volume maintained on the secondary side of our
23 steam generator, essentially making the steam generator
24 adiabatic. The other was run with the steam generator in a
25 hot standby condition.

What we are seeing here is virtually the same

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1 What we are seeing here is virtually the same cold
2 leg mass flow rate. So this difference, at least for the
3 large break over this particular time period, did not
4 influence the blowdown behavior.

5 For reflood, we can imagine there would be a
6 difference simply because of steam binding, and whatnot. As
7 you will see a little bit later in the presentation,
8 however, your intact loop generator does have
9 characteristics in terms of secondary surface area-volume
10 ratios that are approximately correct so that we think we
11 are probably reasonably well-scaled in that respect for
12 reflood concerns.

13 I would like to leave the large-break domain and
14 discuss in the remainder of the presentation similar points
15 for small-break scaling. .

16 (Slide.)

17 On the slide you are about to see, I have not
18 listed all the positive points I listed for the large
19 breaks. I would like to make it clear that naturally we
20 consider that those positive points that we have in the
21 system for large break are also positive points for the
22 small break. That includes such things as energy and mass
23 distribution, resistance distribution, relative elevations,
24 and the ability to do time scale properly.

25 There are some positive points, I think, that need

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1 to be brought out for the small-break considerations. That
2 is that we do have the loop seal elevations proper. While
3 it is not completely clear yet how important that is, we do
4 feel that it is necessary to have the loop seal elevations
5 correct so that if it is a concern, and can be shown to be
6 my analysis which is shown in some of the data, it will be
7 a nonconcern.

8 We do have active broken-loop components which for
9 a small transient, such as a small break, would indeed be
10 important.

11 We do have the proper core length and geometry
12 which, as Dr. Tong alluded to yesterday, is important from a
13 core uncover and boil-off standpoint. I listed one thing
14 here that is not necessarily a positive point, I guess, but
15 it is a plus for the system. Should we find that due to the
16 small sizes of orifices that we have to use to scale break
17 flow, should there be problems encountered there in terms of
18 boundary layer effects or whatnot, we can easily change our
19 break area volume and get the critical flow rate we want,
20 say, in terms of LOFT counterpart testing.

21 I will address that more in a moment.

22 Some of the concerns, naturally, for a small break
23 are the same that you have in a large break. That includes
24 heat transfer, perhaps in a little different light for a
25 small break, in that losses are important, extremely

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es/dd 1 important. Such things as piping thermal time constants are
2 also important.

3 For a number of years, Westinghouse people have
4 brought up the consideration of flow regimes. Your flow
5 regimes are perhaps different because of your fat pipes or
6 your pipes that are of improper length. We have looked in
7 this area. The pump behavior is a concern, particularly
8 from the degradation standpoint.

9 Critical flow is a concern for the reasons I just
10 mentioned. I will talk about what we plan to do and are
11 doing at the present time to look at that.

12 The steam generators, of course, are a concern,
13 especially from slow transients, natural convection, boiler
14 mode environments.

15 Dimensionality, which has come up several times,
16 is obviously important from the standpoint of perhaps
17 natural convection, an uncovered part of the core, or
18 perhaps penetration.

19 DR. PLESSET: There are some absolute numbers --
20 For instance, if you are concerned with pressurized
21 behavior.

22 MR. LARSON: I intend to discuss that.

23 DR. PLESSET: Oh, okay.

24 (Slide.)

25 MR. LARSON: The first thing I would like to touch

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1 on is concerned with external heat losses. The system is
2 insulated with insulation on the external parts of the
3 pipes, the vessel. We have not been terribly concerned in
4 the past about heat losses because of large breaks; the test
5 is over in a rather short period of time. Losses were not
6 thought to be in the area of consideration. When we started
7 to do TMI-type simulations, it became quite apparent that
8 losses were indeed extremely important and, in fact, were of
9 a large percentage of our decay heat.

10 This slide simply shows the comparison both in
11 terms of percent of total power and also absolute values for
12 LOFT and PWR and Semiscale. For PWR values, it was given to
13 me by a man that works for Combustion Engineering. It was
14 taken from a System 80 plant during the start-up phase.

15 I can't at this point in time say that this would
16 be the same value of heat loss for a Westinghouse or B&W
17 plant. I expect it would be in the same neighborhood.

18 The value for LOFT was measured during the L3-0
19 experiment, I think. Our value was measured during one of
20 our TMI simulations.

21 I think you can see the progress here. As the
22 facility size decreases, the heat loss in terms of percent
23 of core power increases by roughly layers of magnitude.

24 The question is what are we doing to try to
25 rectify the heat loss problem? There are a number of

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1 determine where the relative losses are. I don't think it's
2 worthwhile to spend much time on this slide except to
3 indicate that our vessel in the downcomer constitutes a
4 similar part of the loss. Some of the fixes for that are to
5 insulate with greater care the outside of the facility, even
6 if you find that you are still going to have significant
7 heat losses. Some ideas such as using external heat tape
8 have come up. We are currently evaluating that in terms of
9 what it will do to our instrumentation in terms of magnetic
10 field and whatnot. We don't think that will be a problem.
11 We are hopeful we can get this loss down to zero, with the
12 use of heat tape.

13 (Slide.)

14 There is another concern that was brought up.
15 That is piping and time constants. What happens in a length
16 of pipe when the fluid temperature, say, at the inlet
17 changes by a step change; say, for example, what happens
18 during a loss of feedwater; when the fluid temperature at
19 your inlet could change from 540 to 600 degrees?

20 We used a rather simple model conduction
21 calculation and entered the balance on a given volume of
22 fluid which can be representative of any kind of pipe you
23 want, whether it be reactor pipe or Semiscale pipe. It
24 accounted for the proper lengths. Essentially we did the
25 balance to see what the effect of the piping metal mass was

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- 1 on the fluid temperature transient.
- 2 The results are shown in the next slide.
- 3 (Slide.)
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1 They indicate that that appears at this point in time
2 to be a non problem. Take a look at the time scale here.
3 That's over 40 seconds. It shows that both for the Westing-
4 house, which is the curve labeled 1, and the B and W type,
5 labeled 2, the outlet of this length of pipe sees the step
6 changing temperature approximately with the transit time
7 which for a 20-foot length of cold leg pipe is something on
8 the order of 3/10ths of a second.

9 DR. CATTON: What is the limiting factor?

10 MR. LARSON: The metal mass of the pipe. In Semi-
11 scale, we have metal masses that are large. What happens is
12 that it takes a certain amount of time to warm the initial
13 thickness of the pipe up. It turns out to be a non problem
14 here because the differences you see for any significant length
15 of time are less than one degree.

16 I should point out that this was the case for a
17 pipe that's essentially perfectly insulated on the outside,
18 which we do not have at this time. So these results could
19 be a little bit misleading for the system as it presently is;
20 but with the use of improved insulation and heat tape on the
21 outside, it would appear this is a non problem.

22 The next slide simply shows a blow-up of that same
23 slide. I think it is just included to show the relative
24 magnitude.

25 (Slide.)

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1 As I said earlier, flow regimes have been a concern
2 and have been for many years in terms of scaling. Some of
3 the influences that flow regimes can have are that it affects
4 the wetted area which will in turn affect heat transfer. They
5 obviously affect pressure drops. They can affect critical
6 flow depending upon where, for example, the break occurs on
7 the pipe.

8 Some of the treatments that we have been looking
9 at lately are flow regime calculations using bubble type
10 techniques. Eventually when data becomes available from LOFT
11 type small breaks, we intend to experimentally investigate
12 the differences of flow regimes since both the LOFT facility
13 and the Semiscale facility have instrumentation in terms of
14 multi beam densitometers to allow some sort of interpretation
15 as to what the flow regime was.

16 The next slide shows the results of application of
17 the Duckler-Taitel method to a reactor pipe which in this case
18 was a cold leg. It's a 27-1/2 inch pipe. And a Semiscale
19 pipe.

20 The technique used here was to take a mass flux
21 from a PWR calculation and scale that mass flux to that in
22 Semiscale, much the same as the slide I showed of the LOFT
23 versus the Semiscale broken loop.

24 What essentially was done was to take the mass flux,
25 compute from that a mass flow rate for the PWR, divide that

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1 by the power ratio which is a scaling factor to produce the
2 flow rate in the Semiscale. That flow rate in turn was used
3 with the Duckler-Taitel method in calculating what the regime
4 was as a function of time.

5 I think you can see once the system went two phased
6 here, that for a short period of time, both pipes were calcu-
7 lated to be an intermittent which is essentially a sluggish
8 flood flow. There is a slight difference in the flow regimes
9 for the two pipes, from 300 to 400 time frame.

10 A close examination of what is happening here
11 indicates that for the Semiscale pipe, if you translate the
12 Duckler-Taitel method into JG versus JF, you find you are
13 right on the boundary.

14 For the remainder of the periods of time, the
15 velocities for this particular break size were so low that
16 both systems were calculated to be in a stratified type flow
17 regime.

18 Now, I think this is important in that it does
19 show that our pipes are such that we will get separated flow
20 during a slow transient. It is not clear at this point in
21 time that the fact that we do get separated flow in both
22 systems will provide the same response. That requires, I
23 think, some more analysis and is one of the things that we
24 intend to look at in the future. I say that in terms of
25 watability, what the relative water height is in the two

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DD
1 different pipes, how much of the pipe is covered. That in
2 turn will affect heat transfer rates and whatnot.

3 I would also like to point out that similar
4 calculations have been done for other points in the system.
5 This, by the way, was the broken loop pump size break. We --
6 the comparisons actually look a little but better in the other
7 parts of the system, in other words, the intact loop hot and
8 cold legs and the broken loop hot and cold leg (Slide) pumps
9 are indeed a concern primarily from two standpoints, one,
10 degradation, and two, leakage. By virtue of the small size
11 of our pumps, our leakage rates and also by virtue of the
12 number of stringent transients we put them through tend to
13 have seal leakage rates that are quite high. Generally for
14 large breaks we do not make up that leakage. It should pose
15 no problem for small breaks, because through ingenious techni-
16 ques we can trap, weigh, and replenish that seal leakage. It's
17 not easy but we can do it.

18 The primary concern is two phase degradation, small
19 pumps versus large pumps. I think it's a well-accepted fact
20 that a small pump will degrade faster than a large pump.

21 We do not have some treatment for that in that we
22 are buying a new pump for our intact loop that should be an
23 improved design. It will not, however, replicate the PWR
24 pump. It can't. If you use specific speed as a scaling
25 criteria, the new pumps will have a specific speed of on the

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1 order of 1500. A PWR pump, on the other hand, is 5200. There
2 will be differences.

3 We do, however, have plans for a pump testing
4 program. We will quantify the pump performance in all four
5 quadrants, both single-phased and two-phased operations.

6 The next slide shows, I think, the obvious fact
7 that out pumps do tend to degrade somewhat earlier than a
8 large pump. This is a comparison of one of our TMI experi-
9 ments with the data from TMI in terms of a normalized loop
10 flow.

11 The point at which the Semiscale pumps start to
12 degrade is at about a void fraction of 20 percent. We have
13 done some calculations assuming void distributions in the
14 TMI tests and concluded that the big pump will pump rela-
15 tively well until the void fraction is about 40 percent.

16 I should, however, point out that the flow rate
17 from the TMI plant comes from what they call a Gentilly tube.
18 I am not all that confident that it's an accurate measurement
19 for two-phase flow. It may be and is something to keep in mind
20 here.

21 DR. PLESSET: When was that data obtained, this
22 TMI data?

23 MR. LARSON: This data came from what is called the
24 Reactimeter which I am not sure if you are familiar with the
25 nomenclature. It's a small computer that during normal

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1 operation was operating on line and recorded something like
2 25 channels of data, including pump flow, second area pressures,
3 things like that.

4 DR. PLESSET: So we had that available?

5 MR. LARSON: Oh, yes. I think this also was
6 available on strip charts.

7 DR. PLESSET: I didn't know they had that available.

8 MR. LARSON: Yes. There's 25 channels of data
9 that were recorded at like three-second intervals throughout
10 the whole transient.

11 DR. PLESSET: Do you know if they ever made any
12 request for this data from the Reactimeter sheets?

13 MR. LARSON: They?

14 DR. PLESSET: The operators.

15 MR. LARSON: I don't know. I doubt it.

16 MR. ALLEMAN: It was in the room below the main
17 control room.

18 DR. PLESSET: So they would have had it?

19 MR. LARSON: I think there was to be an engineering
20 unit.

21 DR. PLESSET: Thanks.

22 MR. LARSON: Critical flow is another topic of
23 concern from the scaling standpoint. As I alluded to earlier,
24 there are two areas; number one, the small size of our orifices.
25 The data shows there may be a change in the critical mass flow

1 rate as diameter increases. We also are getting down in the
2 regime where the diameter of the orifice is such that pound
3 layer effects are going to be important. There are basically
4 two treatments of that. One, we have test data, particularly
5 from Three Mile Island and experiments we have done; we also
6 have some data from S-02-6, although it's got large error
7 bands on it.

8 We also intend, as an item for the meeting in Wash-
9 ington, D.C., in late July, to do what are essentially cali-
10 bration tests before we run these experiments on the size of
11 the orifices that we plan to use and also that LOFT plans to
12 use.

13 These will be done in the LOFT test support facility.
14 What is hoped here is that we can actually get plots of
15 critical flow rate with stagnation and pressure.

16 The data we got from our Three Mile Island simula-
17 tions, however, is encouraging, I think. This slide shows
18 the comparison of the flow rate we got through our POV valve
19 simulation, I should say, compared to what we calculated with
20 the HEM model using the familiar correction factor of .48.

21 The .48 is not an arbitrary factor. It's a factor
22 we found necessary to use way back in the days of blow down
23 testing in the Semiscale MOD I facility where the .48 was
24 found necessary to account for what was thought at that time
25 to be vena contracta effects.

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1 The .48 was also verified by some work at Los
2 Alamos where two dimensional formulations of our break nozzle
3 were compared with one dimensional formulations. The ratio of
4 those two was found to be about .85.

5 There is an additional bit of evidence for the
6 applicability of that correction factor or contraction
7 coefficient, whatever you choose to call it, in that if a
8 mass balance was done on the system during this particular
9 TMI transient using delta pi's and whatnot, and that was
10 compared to what we calculated with the HEM model, since --
11 through the whole transient the flow was saturated, either
12 liquid or vapor, that you get a .86.

13 Now, I do wish to point out here that this was
14 calculated using either saturated liquid density or saturated
15 vapor density. That may be a consideration when we go to two-
16 phase flows in extremely small orifices. There may be effects
17 not shown here. That is indeed the reason we tend to test
18 these orifices in the LOFT test support facility.

19 The break size for this particular experiment was,
20 by the way, about .3 percent. In other words, .007 squared.

21 (Slide.)

22 Steam generators, as I mentioned, have two
23 potential functions, one being an oversized second area,
24 the second being elevation primarily in the intact loop for
25 a small break. That is important in terms of natural

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1 convection rates. It also affects the operation of the steam
2 generator and the reflux boiler mode.

3 We have two treatments. The near term as I mentioned
4 earlier is perhaps -- one viable thing is to drain the steam
5 generator. We are also attempting to quantify the chances of
6 elevation primarily through the use of RELAP.

7 We are also looking at the operation of the intact
8 loop generator in the reflux boiler mode, since the tubes in
9 that generator are smaller than they are in a PWR. They are
10 .408 inches relative to .775.

11 The calculations we have done to date using a
12 Wallis type formulation indicate that when the system is
13 operating in the reflux boiler mode, that should not be a
14 problem. The velocities are low enough to allow the liquid
15 to fall back into the loop so there will be a counter current
16 flow situation there in the reflux part of the mode.

17 The long-term fix is to provide a new steam
18 generator for the intact loop. We are also looking at the
19 design of that carefully so that we can hopefully get a
20 second area volume that is appropriate; and we are also
21 looking at reducing the second area volume in the main
22 generator, if we can, through the use of filler pieces.

23 I think it is important at this point in time just
24 to indicate on this slide what the relative second area side
25 heat transfer areas are in our facility.

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1 (Slide.)

2 That is relative to that in a PWR. Granted, as you
3 will see on the next slide, our second area volume in the type
4 2 generator is large; but if there is not a heat capacity
5 effect, what we see here is that this is a function of
6 normalized level. We do have a scaled heat transfer surface
7 area. In fact, the intact loop generator is not that far off.
8 I should point out that this type 1 generator actually simu-
9 lates three PWR generators, so that the surface area was
10 divided by three to make this comparable.

11 DR. CATTON: What about the flow rate on the
12 primary side?

13 MR. LARSON: In the horizontal loop or in the tube?

14 DR. CATTON: Vertical; in the steam generator.

15 MR. LARSON: As I just said, in terms of the
16 reflux boiler mode operation, we will get counter current
17 flow based on some assumptions in terms of what the steam
18 generation is in the core.

19 The velocities in general are lower than in the
20 PWR.

21 DR. CATTON: Your heat transfer coefficients on
22 the primary side might be different.

23 MR. LARSON: That is possible.

24 DR. CATTON: So the surface area doesn't have quite
25 as much meaning?

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1 MR. LARSON: That's true.

2 DR. CATTON: You need to look at the whole problem,
3 not just pick one side.

4 MR. LARSON: You need to have the right velocities.
5 That's correct.

6 (Slide.)

7 This slide simply gives a comparison of some
8 normalized values and some ratios that are important in
9 terms of scaling water volume and whatnot. It shows quite
10 obviously the oversized second area -- or the type 2 generator.
11 It does indicate that the type 1 generator, from a heat
12 capacity standpoint, is quite close. In other words, it's
13 the right amount of volume -- or at least close to it -- to
14 simulate 3PWR generators.

15 In terms of surface area over primary volume, we
16 are quite close. It also shows the relative elevations of
17 the type 2 and type 1, the type 1 being short.

18 Dr. Plesset mentioned pressurizer surge line a
19 moment ago.

20 (Slide.)

21 Particularly in the TMI transients we have done,
22 there has been a lot of concern on both what the size of the
23 surge line was and also the hydraulic resistance. The simple
24 treatment for that was when we did the TMI simulations, we
25 had two requirements: One was to incorporate the dog leg in

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1 relative elevations; two, we wanted to get the right scaled
2 resistance which I think we accomplished.

3 (Slide.)

4 This slide shows some pertinent facts regarding
5 the surge line relative to the B and W designed surge line.
6 Our surge line is about .37 inches in diameter, or was. The
7 PWR, on the other hand, is about 8-1/2 inches. As far as
8 the resistance, we calculated the resistance for the PWR
9 surge line and scaled it up by the volume we show, squared,
10 and got a value of about 6800 in English units.

11 We did the same thing for our surge line to
12 determine what sizes we needed, and we got about 6200. This
13 value was measured and found to be somewhat larger. We were
14 closer to this. I think the value is about 6800 or so.

15 Much the same as we test our loop components before
16 we operate the facility, we endeavored to get as close to the
17 scaled resistance value as we could on this particular simu-
18 lation.

19 In terms of looking at the size of the surge line
20 and in terms of surface tension effects and what that does in
21 terms of drain rates, we have looked at a dimensionless dia-
22 meter concept in the flooding theory.

23 That, for the PWR, was about 133. For our surge
24 line, it was about 5.8. As near as we could find in the
25 literature, we accepted a critical value of something like

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1 .814 for those sizes where surface tension effects and
2 bridging are important enough to allow a line not to drain.

3 So at least based on the available theory, it
4 would appear that that line size is not sufficiently small
5 to cause those kinds of problems. We have also looked at --
6 from a Wallis correlation standpoint -- what the critical
7 gas velocities and what the real superficial velocities in
8 the surge lines were for a given state in the TMI transient.

9 What we did was look at the case where the plant
10 was operating at roughly 1000 psi. We said, okay, the HEM
11 model is a reasonable technique by which to calculate the
12 flow rate; translated that, assuming that there was a continu-
13 ity -- continuity of mass; what superficial gas velocity was
14 required in the surge line to create cold current flow.

15 I think as you will see, the critical gas velocities
16 in the PWR was about 13 and about 2 in our surge line.
17 However, the real velocity in the PWR would be about 42, and
18 ours would be about 20. It would appear that in both cases,
19 we have cold current flow. As I said, this is based on a
20 Wallis correlation. 8-1/2 inches may be a diameter large
21 enough where you should use the Kutelatzze criteria.

22 DR. PLESSET: What situation are you going to apply
23 this to? In the hot leg, say you have steam, and you have a
24 filled pressurizer or steam is going into the pressurizer?

25 MR. LARSON: Exactly. That was the assumption.

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1 DR. PLESSET: Yes.

2 MR. LARSON: Those were the ground rules.

3 DR. PLESSET: I doubt if these criteria are

4 adequate or appropriate. The absolute length of the line

5 will make a difference in the rates. Am I right in that?

6 These criteria are not appropriate ones. So I

7 would recommend that you look at that again.

8 MR. LARSON: We have had ideas of looking at this
9 experimentally.

10 DR. PLESSET: Oh, that's all right. Do that.

11 MR. LARSON: We haven't proposed this yet, but

12 I think it would be a reasonable thing to do.

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1 DR. CATTON: You don't have that in Semiscale?

2 DR. PLESSET: You could do that on a pipe
3 somewhere.

4 Also, I think that since gravity enters, the
5 absolute heights are important; don't you think so?

6 DR. WU: For some nondimensional number.

7 DR. PLESSET: Maybe some nondimensional number.

8 MR. LARSON: We did maintain the proper heights in
9 terms of the dog leg.

10 DR. PLESSET: That part. I think the whole
11 geometry of the whole line, its length, the height of the
12 pressurizer relative to the hot leg. So it is not only the
13 dog leg that's important.

14 DR. ROSZTOCZY: Dr. Plesset, this has been looked
15 at recently in terms of the various PWR vendors. They have
16 been looking at the problem.

17 As I understand it -- and I think it is the same
18 as has been done here -- the approach is that should you
19 have, number one, the break in the pressurizer, an opening
20 of the valve on a pressurizer, then the flow in the surge
21 line, the steam flow in the surge line, going into the
22 pressurizer -- the question is under what circumstances
23 would this steam flow prevent water from leading back into
24 the system.

25 DR. PLESSET: Sure.

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1 DR. ROSZTOCZY: The cross-section of the piping is
2 usually uniform, so there is no difference there. The steam
3 velocities are very high. We know the flooding occurred in
4 the end of the pipe. If there is any water, it is pushed
5 into the pressurizer. There will be no drainage back.

6 What is being done here is a simple flooding
7 check, let's say, at the end of the pipe or at the smallest
8 cross-section, wherever it exists. Usually there is a cone
9 at the entrance which has holes in it. It has a smaller
10 cross-section than the pipe itself.

11 You do a simple flooding check. You are looking
12 at the gas velocities in that cross-section and see if, with
13 those gas velocities, is it possible to penetrate any water
14 against it.

15 The answer is what is shown there. The vendors
16 are coming up with the same answer. The gas velocities are
17 significantly higher. How much would be needed to hold up
18 the water; therefore, no water penetration is critical. So
19 for those breaks, independent of the U-2, most of the plants
20 do not have a U-2 type of surge line. They have a straight
21 surge line. Even with the surge line, they do not predict
22 that.

23 DR. PLESSET: Well, I would remark — you talk
24 about a straight pipe. I am told that there's a horizontal
25 run of 30, 40, 50, 60 feet in some of these installations.

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1 Just a couple of days ago, a man was very proud
2 when he told me this horizontal rod was 60 feet. He seemed
3 pleased with that. I think this is a very important aspect
4 of this gas liquid flow, the fact that they are enormously
5 long. I think if we don't really look at that, we may be
6 neglecting something quite important.

7 DR. ROSZTOCZY: Then the calculations are done to
8 predict the flow, the flow that you are -- have been going
9 to use to check on, then those calculations are done with
10 the proper flow resistance of the system, so if there is six
11 foot of pipe, the calculations -- the original calculations
12 are done taking full account of the 60-foot length of the
13 pipe, appropriate flow resistance corresponding to this.
14 Obviously if those are something different, the pipe would
15 be shorter.

16 Once you have that flow, then you can take the
17 calculated flow which accounted for this and compare it
18 against the flooding condition you see, the penetration, and
19 arrive at the appropriate conclusion.

20 DR. PLESSET: Are these conclusions meaningful?

21 DR. ROSZTOCZY: Yes.

22 DR. PLESSET: Are you sure?

23 DR. ROSZTOCZY: Yes.

24 MR. ALLEMAN: If it floods for a short length, it
25

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1 would flood for the long one.

2 DR. PLESSET: I think rates are effective.

3 DR. WU: The rates may be affected by change of
4 the flow configuration such as there is a stratified
5 concurrent flow. When you increase the pipe length because
6 of the increased friction, it might generate the flood
7 flow. When that occurs, the breathing in the upper layer of
8 this counter-current, might be peaked.

9 Also the strain above the outlet of the vertical
10 tube inside of the pressurizer also is scale dependent. We
11 have tested some of the spring size. It can be very
12 different. The law of basic physics dictates on a change of
13 the major flow configuration and type of flow.

14 Those are some of the things you have to
15 consider. It might be a beneficial thing to consider.

16 DR. ROSZTOCZY: The picture I am getting from this
17 type of calculation is that we are not at the flow velocity
18 at all of the things you are describing could happen. We
19 have an order of magnitude higher in the velocities. There
20 is no chance for any of it to come back.

21 So if you take the picture and increase the gas
22 velocities by a factor of 10, then it will be somewhat
23 different.

24 What we were getting here — I think it shows 42
25 against 11. So a factor of 4. Calculations that we get

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1 from the vendors were showing also factors of two, three,
2 four times higher than that.

3 So I think maybe a factor of 10, but a factor of
4 two, three higher velocities where the transmission would
5 take place.

6 If the velocities get smaller, if those two
7 numbers would be comparable numbers, it is rather complex.
8 It is very difficult to predict exactly what will happen.

9 DR. CATTON: At the outset, the gas velocity is
10 zero. Then there is a transition to the point where this 42
11 feet per second. I think one of the questions is how long
12 does it take to get there? If that's a short time for this
13 flow to establish, then I think you are right.

14 DR. PLESSET: It depends on the rate of
15 temperature buildup, the stream.

16 DR. WU: That's right.

17 DR. CATTON: If it takes a long time to establish
18 the flow, then there is a consideration.

19 DR. ROSZTOCZY: Yes. Obviously there is a
20 transition period that you go through when you get into the
21 high velocity zone. If you wait long enough, there will be
22 a transition again and again.

23 DR. CATTON: The length of time it takes for the
24 transition to occur, and what kind of things are happening
25 during that period, and is it important --

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1 DR. PLESSET: I think we pointed it out. I think
2 we are at the point where we need further examination.

3 DR. CATTON: RELAP won't do it.

4 DR. ROSZTOCZY: These are all calculations done on
5 the side.

6 DR. PLESSET: Well, go on.

7 MR. LARSON: Okay. We are almost through.

8 The last topic I would like to briefly address is
9 that of dimensionality. As I said earlier, our system is
10 largely 1D by virtue of the large L/Ds. We have seen
11 multidimensional phenomena in our core. In the case where
12 we injected it into our upper plenum, there was a
13 significant amount of dimensionality in terms of the core
14 cooling.

15 . However, I think we will largely treat this
16 analytically. It is a difficult phenomena just to do hand
17 calculations with regard to, I think. It is something that,
18 from a scaling standpoint, we will have to evaluate in the
19 future.

20 (Slide.)

21 In summary --

22 DR. CATTON: I didn't quite understand that.
23 Would you put that slide back?

24 Would you repeat what you just said?

25 MR. LARSON: In what regard? In terms of the

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1 analysis?

2 DR. CATTON: The 1D versus 2D.

3 MR. LARSON: What I said was we ran an experiment
4 in the MOD 1 facility which also was construed to be largely
5 one dimensional by virtue of the large L/Ds. We have run an
6 experiment where we have injected ECC water into the upper
7 plenum. We have noticed two-dimensional cooling
8 characteristics, even though the core is left out. So there
9 are two-dimensional effects in the system.

10 Whether or not they are typical of a PWR — they
11 are probably not, but there is a potential for
12 two-dimensional effects.

13 I think in the gross sense, the effects of
14 dimensionality will have to be evaluated via a
15 multidimensional code.

16 DR. CATTON: If the interest is in a phenomena
17 that may be multidimensional, why do you run a test?

18 MR. LARSON: Are you referring to UHI?

19 DR. CATTON: Just anything in particular.

20 MR. LARSON: Well, let's take UHI as an example.
21 When the scaling for UHI was agreed upon, it was not, in my
22 estimate, the multidimensionality of the effect that was
23 thought to be the important thing. What people were
24 interested in was the regimes of the UHI and, in essence,
25 the gravity dominated situation such as draining that was of

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1 interest.

2 So from that standpoint, it was of interest to
3 have proper core limit.

4 DR. CATTON: Draining of the upper head?

5 MR. LARSON: Yes.

6 DR. CATTON: Isn't that also a multidimensional
7 problem?

8 MR. LARSON: It could be, but it is a gravity
9 dominated situation.

10 DR. CATTON: It seems to me if you have
11 multidimensionality, the lateral stands are also important.
12 If you restrict --

13 MR. LARSON: Full scale.

14 DR. CATTON: I don't think so.

15 DR. PLESSET: Dr. Tong?

16 DR. TONG: I would like to make a comment. In the
17 small break, the flow rates are usually show. I see a
18 three-dimensional effect being much less than in a large
19 break. However, the UHI is an exception. I wish that small
20 tests not talk about UHI for the time being, because most of
21 the plants we are talking about have UHI.

22 Getting to UHI, we have to systematically study
23 how this one-dimensional -- three-dimensional effect
24 including UHI.

25 UHI simplified -- it is a different question

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1 comparing it with ordinary plant small break studies.

2 MR. LARSON: I would agree with that.

3 DR. PLESSET: Thank you.

4 MR. LARSON: In summary, I think at least the data
5 we have gotten to date leads us to believe that volume
6 scaling is appropriate. We knew that distortions were
7 present in all systems that were scaled. Ours is not
8 different.

9 Most of the things we have looked at to date are
10 not items that are, from a code standpoint, unmanageable.
11 They are accounted for, so from that standpoint, the data
12 should be useful from code benchmarking standpoints. We
13 also feel the data will be useful for phenomena
14 identification. We also need the data to help quantify a
15 log of distortions we discussed.

16 In general, our small break distortions aren't
17 fully quantified. What I endeavor to do is give you an idea
18 of some of the things we are presently looking at, hopefully
19 to impress upon you that we are working on what we consider
20 to be an extremely important consideration.

21 (Slide.)

22 In the small break area, we have a joint effort
23 established with the LOFT program and also we are getting
24 help from the Australian Government.

25 DR. PLESSET: What are they doing?

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1 MR. LARSON: They have some free manpower. It is
2 not exactly clear what they are going to do yet, but they
3 have the ability to run codes, engineering analysis.

4 DR. PLESSET: They also have a reactor that's not
5 being used. Maybe we could use that.

6 (Laughter.)

7 VOICE: We have no reactors, but we have the
8 manpower.

9 DR. PLESSET: I thought you had one that was
10 essentially complete. Is that not correct?

11 VOICE: We have one completed, but not yet taken
12 into service.

13 DR. PLESSET: So it is available?

14 VOICE: It is available.

15 (Laughter.)

16 DR. CATTON: Available for what?

17 MR. LARSON: Some additional scaling work is
18 undoubtedly required for loss of feed water type tests and
19 anticipated transients. We intend to document this work
20 much the way we documented the three scaling efforts we did
21 for the large break testing.

22 I think there needs to be an assessment of the
23 effects of dimensionality.

24 That concludes my presentation. If you have any
25 questions, I think all of my scaling friends are here, and I

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1 would like to direct the questions to them.

2 (Laughter.)

3 DR. PLESSET: I think we expressed most of our
4 concerns and interests. We appreciate your presentation.

5 We have two more presentations to do. I
6 guaranteed we will finish in time so that people can make
7 that flight that they need. One presentation is by
8 Mr. Johnsen on some TMI work on Semiscale. The last will be
9 on a test plan and upgrade to perform transient tests in
10 Semiscale.

11 I propose we take out lunch break now and return
12 as close to 1:00 o'clock as we possibly can.

13 (Whereupon, at 12:15 p.m., the meeting was
14 recessed, to reconvene at 1:00 p.m. this same day.)

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AFTERNOON SESSION

(1:05 p.m.)

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3 DR. PLESSET: I will call the meeting back into
4 session.

5 We made a slight shift in the order. The first
6 presentation will be on plans for FY '80-'81 in Semiscale by
7 Mr. Olson.

8 MR. OLSON: Your handouts have got a more extensive
9 presentation which kind of takes you through the point where
10 we were last year to today. In the interests of brevity, I
11 will just concentrate on our current scope as we see it for
12 FY '80-'81. (Slide.)

13 DR. PLESSET: I apologize on your name. The agenda
14 had it spelled wrong.

15 MR. OLSON: I noted that. It was pretty close.

16 Basically we have discussed in the last couple of
17 days the emphasis on small break testing. That emphasis
18 carries through into Semiscale. We are planning a small break
19 testing effort covering a period from the end of this month
20 into March of 1980.

21 In Semiscale a small scale testing effort is
22 basically a two-stage process. The first stage has been
23 closely coordinated with LOFT in terms of both the supportive
24 and comparative role. Tarrell Samuels presented the plan for
25 LOFT. Mr. Harvego will give the Semiscale plans.

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1 Basically we have taken the same concept of a
2 certain number of tests and scaled those tests or designed
3 those tests using the same criteria.

4 In this manner, we are basically performing tests
5 in which we will be able to directly relate Semiscale and LOFT
6 data and again as we did in the large break situation, give
7 some initial indication on the effects of physical scale
8 between the facilities in the small break phenomena area.

9 The second stage of small break testing is not yet
10 completely defined. But it will address issues that have
11 been raised yesterday by Dr. Tong relative to some separate
12 effects type investigations and will address some of the
13 outstanding issues that Zoltan talked about this morning.

14 In designing these tests, it is our intent to
15 specifically address scaling issues in each of those tests and
16 hardware requirements in each of those tests prior to the
17 time that we do conduct those tests so we can assure that they
18 are completely meaningful tests in Semiscale.

19 In between those two stages of small break testing
20 we will be repeating S-0-76 with the new downcomer insulator
21 installed. That has been designed, fabricated. It has been a
22 state of the art effort in the design and fabrication
23 activities.

24 That hardware is not complete, however. We will be
25 ready to perform these tests as soon as we go through the

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1 first stage of small break testing.

2 We are also, as a result of Three Mile Island
3 emphasis, planning a scoping loss of feed water test in
4 Semiscale. Mr. Harvego will discuss the test rationale,
5 objectives, and expected results — basically the philosophy
6 with which we are approaching these tests.

7 These tests will be performed in a period from
8 April to October basic conforming to the general guidelines
9 trying to obtain more general results by September of 1980.
10 We are at a minimum in conducting these tests intending to
11 include those hardware modifications which will eliminate some
12 of the nontypicality that was discussed this morning,
13 specifically the external insulation problem, the internal
14 insulation, both in the downcomer and in the vessel itself, as
15 well as installation of the type-two steam generator, the PWR
16 steam generator, to eliminate those nontypicalities that we
17 are aware of.

18 Past that point then, we have scheduled the UHI
19 sensitivity studies which have been delayed significantly
20 because of the lack of capability for the UHI calculation.
21 Zoltan talked about that this morning.

22 To this point, our schedule is basically that which
23 was presented in December to the NRC with modification because
24 of the emphasis of the small breaks and feed water testing.
25 We are dealing with a projected scope which involves

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1 installing -- designing and installing new hardware systems
2 in Semiscale which you have been presented information on in
3 past meetings relative to supplemental budget required for
4 Semiscale.

5 These are the closed-loop secondary system and the
6 two-by four loop simulation.

7 What I have on this schedule is a minimum -- or a
8 schedule based on time phases of testing. This schedule can
9 be accelerated or modified depending upon the priorities.
10 This is our current plan.

11 We are currently involved in the preliminary central
12 design of both the closed-loop secondary systems and the
13 two-by-four loop simulation. Concurrently, I would like to
14 assure you that while we are doing those preliminary designs,
15 which is necessary in terms of designing standpoint, which is
16 a fairly long-term activity, we are addressing scaling issues
17 in terms of how well we can scale closed-loop secondary --
18 what is actually needed in the closed-loop secondary in terms
19 of performing meaningful tests for Semiscale.

20 Those analyses and evaluation we intend to complete
21 before we get to the point where we actually have the
22 hardware. What we are intending to do is go through the
23 cycle where we actually address the specific scaling issues,
24 address the specific tests that can be meaningfully performed
25 in Semiscale and factor that into the design before we go off

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1 into the hardware.

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2 Basically that is a condensation of what I was
3 going to present. I think Ed will be presenting now the
4 details of the planning for the small break and feed water
5 tests in Semiscale.

6 DR. PLESSET: Thank you. I see no questions.
7 Thank you very much. Harvego.

8 MR. HARVEGO: Good afternoon. I am Ed Harvego. As
9 Mr. Olson said I will be talking about the future Semiscale
10 test and schedules.

11 Specifically, I will be talking about our plans for
12 the upcoming small break test series in the final stages of
13 development. I think it will become apparent as I get into my
14 discussion that they are very closely tied to the LOFT small
15 break tests which were discussed yesterday.

16 I also will be discussing the rationale for future
17 loss of feed water tests and the plans for implementing these
18 tests; and finally, I will be talking about the methodology
19 which we have developed for defining as yet undefined
20 operational transients which could be run in Semiscale.

21 (Slide.)

22 This is the schedule for the next couple of years.
23 I think I will just briefly touch on this as Danny has already
24 mentioned it. The objective or our intention in this
25 schedule is to provide LOFT with immediate information in

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1 approximately two to three months on the tests that they are
2 going to be running and then to provide additional information
3 on small break tests within a period of about six months.

4 You will notice on this schedule that we have
5 allowed additional time for some optional tests which I will
6 be talking about in -- a little bit later.

7 The other point is that if we stay with this
8 schedule, we will be beginning our loss of feed water tests in
9 April and continuing those until the fourth quarter of 1980.
10 The operational transients test series would then begin in the
11 second quarter of 1981. (Slide.)

12 To begin the discussion of the small break tests,
13 I would first of all like to point out some of the reasons
14 why we feel these are important tests to be running in
15 Semiscale. First of all, recent events -- specifically TMI --
16 have pointed out the importance of probabilities and the need
17 to understand the causes and consequences of small break
18 accidents.

19 Dr. Rosztoczy mentioned that in his talk also. We
20 feel there's limited data available on small breaks. Prior to
21 TMI, Semiscale ran two small break tests. One was in the
22 MOD-1 system; the other was in the MOD-3 system.

23 During and subsequent to TMI we did a total of eight
24 tests in support of the Three Mile Island incident, and we
25 feel that those tests provided important insight into a lot of

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1 phenomena that actually occurred in the PWR.

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2 We feel that this type of information, or this type
3 of testing, is important to address a wide range of concerns
4 relating to small breaks. We also feel there is a necessity
5 for experimental data for the codes and benchmarks. I am
6 specifically talking about experimental data on the potentials
7 for core uncovering, the influences of core uncovering on the
8 thermal hydraulics and temperature response of PWR cores.
9 (Slide.)

10 Therefore, the objectives that we have identified
11 for the small break tests are first of all, to identify the
12 important thermohydraulic behavior associated with a small
13 break accident and the potential effects on the outcome of
14 that transient.

15 I have also mentioned we are closely tied to the
16 LOFT experimental program; we are also closely tied to the
17 PWR audit calculations.

18 Therefore in these tests, we expect to compare
19 results from Semiscale and LOFT and hopefully identify some
20 of the scaling influences on the important thermohydraulic
21 behavior we observed there.

22 Yes, Dr. Catton?

23 DR. CATTON: I believe the B&W people have indicated
24 that they felt in many respects that the small break was a
25 quasistatic process, basically a series of steady-states.

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1 Also, the Michaelson analysis was performed as if
2 he was looking at a series of steady states. It is very
3 interesting if one of the things that would come out of what
4 you are doing is either to prove or disprove that. Maybe we
5 don't need a new code.

6 I see you have code evaluation; but are you going to
7 look into that -- that aspect of it?

8 MR. HARVEGO: What we have in mind is we have -- the
9 PWR audit calculations, we will have pretest predictions for
10 Semiscale. We will have pretest predictions for the LOFT
11 experiments. What we hope to do is tie these together,
12 identify the important semihydraulic behavior these codes are
13 predicting for each of these -- for the LOFT and Semiscale
14 experiments, and for the PWR; and hopefully use that
15 information along with the experimental data that we obtain
16 to address some of the question about -- in terms of how well
17 can the codes predict certain phenomena.

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1 DR. CATTON: That wasn't the question I asked, but
2 I understand the answer.

3 MR. HARVEGO: Did I answer the question then?

4 DR. CATTON: Well, you did. You essentially
5 answered it no.

6 MR. HARVEGO: I think what I am saying is that if
7 in fact we find out that it is a steady-state problem, and
8 you don't need big codes, hopefully then we will recognize
9 that in our experiments.

10 DR. CATTON: Well, see, I would think that maybe
11 you would do some thinking and calculations sort of back of
12 the envelope beforehand.

13 MR. HARVEGO: I think we have. I will get into
14 that as I continue along.

15 (Slide.)

16 We have five small break tests presently planned
17 for the small break series. The first one is a two and a
18 half percent break corresponding to a four-inch break in a
19 PWR. For this test, we are going to be using the initial
20 conditions from the PWR audit calculations which were
21 performed by our code assessments group. We will be
22 configured in our normal MOD 3 configuration.

23 The things we are going to be looking at in this
24 particular experiment are such things as the break-flow
25 characteristics; Tom Larson already addressed the scaling

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1 concerns in his presentation regarding breakflow. We are
2 also going to be looking at system depressurization
3 characteristics; and finally, the potential for this system
4 depressurization and how — what potential this system
5 depressurization characteristic would have on the potential
6 for core uncovering in both the PWR audit calculations and in
7 calculation that we have been performing in trying to
8 address Dr. Catton's question right now.

9 We are looking — we saw — our calculations
10 indicate that we will get core uncovering in the semiscale
11 facility. We also saw those in the PWR audit calculations;
12 so that's the type of thing that we are going to be trying
13 to address in this first test.

14 The second test we have defined is intended to
15 identify differences in initial conditions and LOFT
16 geometry, differences between the LOFT configuration and the
17 PWR audit calculations.

18 Therefore, this test would again be a two and a
19 half percent break test. The initial conditions would be
20 that from LOFT test L3-1 which was talked about yesterday.
21 We would be -- have our loops configured to represent as
22 closely as possible the LOFT loop configurations;
23 specifically, we are modifying the pump section, the intact
24 pump loop section leg to simulate the elevation effects
25 between the cold leg center line and the cold section invert

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1 so that it would be representative of the LOFT
2 configuration.

3 We are also planning on developing out the broken
4 loop hot leg to give us a break configuration more
5 representative of the non-communicative nature of the LOFT
6 configuration.

7 The third test we have planned is an optional test
8 in that if we don't find significant difference between the
9 test designed to simulate the audit calculations and the
10 second test which was designed to simulate the LOFT
11 configuration and initial conditions, then we probably
12 wouldn't run this test. However, if we do see significant
13 differences, then we want to try and identify whether those
14 differences are caused by the initial conditions,
15 differences in initial conditions between LOFT and
16 semiscale, or LOFT and PWR, whether those are due to
17 differences in the LOFT configuration.

18 Therefore, this third test would be a two and a
19 half percent break; but in this case, the initial conditions
20 would be those from the PWR audit calculations, but we would
21 maintain the semiscale system in the modified LOFT
22 configuration.

23 The fourth test we have planned is the smallest
24 break that we would be running. This is a .16 percent
25 break, corresponding to a one-inch break in a PWR. For

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1 this test, the initial condition will be from the PWR audit
2 calculations; and we will again reconfigure it in a MOD 3
3 configuration.

4 Again in this test, we will be looking at such
5 things as the breakflow characteristics, scaling effects on
6 breakflow characteristics, system depressurization, and also
7 we will be looking at natural circulation effects. Both our
8 initial calculations and the PWR audit calculations indicate
9 that we will go into a natural circulation mode. So we will
10 be looking at the effects of natural circulation, or the
11 phenomena affecting natural circulation and will also be
12 looking at the potential for recovery from the natural
13 circulation mode once it is stabilized in that condition.

14 The fifth test is again another optional test. If
15 we did run this, this would be a 10 percent break. The
16 initial conditions would be for an S-07-10 which was the
17 small break test run in a MOD 3 configuration. We would be
18 configuring the MOD 3 -- we would have the MOD 3
19 configuration with upper head injection.

20 If we ran this test, we would be looking at the
21 potential effects of ECC injection into the upper head on
22 breakflow characteristics and overall small break phenomena.

23 MR. MATHIS: Pardon me just a minute. Zoltan, you
24 mentioned this morning you had serious questions about the
25 high priorities given to this program. Is this consistent

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1 with your desires, needs?

2 MR. ROSZTOCZY: Yes. The program basically has
3 been developed by EG&G and RES. It has been presented to us
4 back a few weeks ago. We reviewed this in some detail. We
5 had a few comments on it. In general, we found that the
6 program that we developed was very responsive to the request
7 that we gave them.

8 Some of the details, in the individual tests, we
9 had some comments. I can't recall all the details. Some of
10 them indicated that there might be some tests that might not
11 be needed. One additional one might be needed here or
12 there. I believe those comments have already been
13 incorporated into the program that is being presented here;
14 is that correct?

15 MR. HARVEGO: Yes.

16 MR. MATHIS: Fine. Thank you.

17 MR. HARVEGO: I mentioned that we had the option
18 for running additional tests.

19 (Slide.)

20 I added this slide. I don't believe this is in
21 your handouts, but I did want to address this. We are
22 considering running additional tests primarily because of
23 comments from licensing and NRC research.

24 Specific areas we think we might want to look at
25 on the phenomena are, for example, we were thinking about

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1 running a duplication of one of the small break tests in
2 which we left the pumps running to investigate the phenomena
3 of performing in effect a separate effects test to
4 investigate effects of pumping — or the operation on small
5 break phenomena.

6 We are also looking at the potential of running a
7 number of natural circulation tests, specifically natural
8 circulatio. tests involving non-condensable gasses in the
9 primary system, reflux natural circulation as a fallback.

10 Not listed on here, but I think also under
11 consideration, is running a hot leg small break test.

12 (Slide.)

13 The results we expect to get from these tests are
14 first of all, we expect to see core uncoverly in the larger
15 break tests. As I mentioned, our calculations, the NRR
16 calculations. the PWR audit calculations indicate we will
17 get core uncoverly. We will also be looking at natural
18 circulation phenomena.

19 Again calculations have indicated that the system
20 will be pressurized in the smallest break to something
21 around the second area site pressure and then go to a
22 natural circulation mode.

23 We will be looking at second area side influences
24 particularly on the natural circulation mode. We will be
25 looking at such things as elevation effects, steam

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1 generator elevation effects, and the potential effect of
2 second area side heat transfer on the natural circulation
3 phenomena.

4 Finally, we will be looking very hard at scaling
5 effects, particularly scaling effects between LOFT and
6 semiscale since we are very closely tied — the two programs
7 are very closely tied together on this series. We expect to
8 get a lot of information on scaling effects in both systems
9 on the overall thermohydraulic response.

10 (Slide.)

11 I would now like to turn to our plans for the
12 future loss of feed water test series. Again, to begin
13 with, I would like to give you some indication of why we
14 feel the loss of feed water test series should be the first
15 operational transience test series run in semiscale. The
16 first is relative probabilities.

17 At the time WASH-1400 was written, operational —
18 or operating data indicated that we could expect an
19 interruption in normal feed water approximately three times
20 per reactor year of operation. I have also got information
21 that was given to me by our reliability group which
22 indicates that the reliability of power-operated relief
23 valves, that class of valve, is something like — the
24 reliability is something on the order of eight failures per
25 one thousand actuations.

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1 If you assume every interruption in feedwater
2 resulted in an increase in the pressure and the actuation of
3 a relief valve that would indicate that the probability of a
4 small break being initiated by a loss of feed water incident
5 or a transient is something on the order of .018 or one in
6 every 50 reactor years of operation.

7 This, to me, is a pretty significant
8 consideration. The second thing is that because the
9 probabilities are relatively high for this type of
10 transient, there has been quite a bit of work in the area of
11 failure and event tree analysis. The event tree is fairly
12 well-defined. I will get into a minute the fact that we
13 feel this event tree analysis provides us with a good method
14 of defining future tests in semiscale.

15 Finally, as with all operational transients,
16 including loss of feed water, there is a significant lack of
17 experimental data in this area to help us in verifying the
18 engineering analysis and in identifying potential
19 safety-related phenomena that the codes might not be able to
20 predict.

21 (Slide.)

22 So the objectives for the loss of feed water test
23 would be first of all to investigate the important
24 thermohydraulic events associated with the single initiating
25 event, assuming that all other systems function as designed.

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1 The second thing — this is where we feel that
2 Semiscale can make a major contribution — is the
3 investigation of potential common mode — or multiple common
4 mode failures. I will discuss that more a little bit
5 later. As I mentioned, we would also be utilizing this
6 experimental data in our code development activities.

7 (Slide.)

8 The plan that we propose is that for the loss of
9 feed water test, every test in that series would be
10 initiated with a loss of feed water event. We would then
11 use an event tree approach in that we would begin looking at
12 multiple failures, common mode failures, and increase the
13 severity of the transient until we reached a transient which
14 we call a limiting case, where there is a potential for core
15 damage in a nuclear plant.

16 (Slide.)

17 To be more specific, I will show you what we have
18 in mind. I have extracted this event tree for a loss of
19 feed water accident from WASH 1400. Basically what it shows
20 is that the event is initiated with a loss of feed water.
21 Each of these branches defines either the success,
22 cooperation of a particular system, or failure of a
23 particular system to function. You sequentially progress
24 through the various system operations which have to occur.

25 At the end of each of these paths, we have

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1 identified whether or not that particular path would lead to
2 core damage.

3 As an example, just to clarify this, in this case,
4 this top branch represents the successful operation or a
5 reactor scram. The bottom branch represents the failure of
6 the reactor protection system to scram.

7 Therefore, this entire category, or this entire
8 sector of that event tree, defines a general category of
9 transients, defined as anticipated transients without scram.

10 What we have in mind for our tests is to begin
11 with a baseline test which would be the initiation of the
12 event, but the successful operation of each of the reactor
13 systems.

14 This is the same type of test that could be
15 performed in LOFT.

16 We hope from this baseline test to utilize the
17 information obtained from our results and compare those with
18 similar tests that would be performed in LOFT.

19 Where we expect to make a significant contribution
20 to this general category of transients is that in — since
21 we are not a nuclear facility, we are not limited by the
22 safety requirements of a nuclear test facility.

23 Therefore, we feel like that we can look at more
24 severe — we are more flexible in that we can look at more
25 severe transients. So taking this information, after we

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1 compared it with results from WALP, we look at multiple
2 failure modes. We can look at things such as the loss of
3 off-site power leading to the interpretative reactor pumps.

4 We can look at delayed auxilliary feed water
5 flow. We continue to increase the severity of
6 these transients until we reach a limiting case such as the
7 one shown here where we had a loss of normal feed water --
8 loss of auxilliary feed water in spite of the other systems
9 functioning as designed. The path still led to core
10 damage.

11 It's interesting to note that in our Three Mile
12 Island testing -- which I think Greg Johnsen will have a few
13 minutes to talk to you about -- we actually followed one
14 of the limiting paths in this event tree.

15 . In Three Mile Island, the event was initiated by a
16 loss of feed water; there was a successful scram in the
17 core. However, the auxilliary feed water flow was at least
18 delayed sufficiently to place us on this path.

19 The power-operated relief valve opened. However,
20 it failed to reclose. Although there was ECC injection, the
21 event tree indicates that that particular event led to core
22 damage.

23 So the results that we expect to get from our loss
24 of feed water test series are first of all, we are going to
25 try to identify the important thermalhydraulic behavior

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1 associated with the loss of feed water transient, including
2 the possibility for multiple common mode failures.

3 We will be using -- secondly -- the Semiscale
4 instrumentation, the extensive Semiscale instrumentation to
5 help us identify the important thermalhydraulic phenomena
6 that are going on and hopefully be able to relate these in
7 some way to a phenomena occurring in a PWR; and in that way,
8 identify the capabilities and limitations of the PWR
9 instrumentation in terms of defining particular condition or
10 particular condition or state of the reactor at various parts
11 in the transient.

12 Finally, we will also be looking at our codes in
13 terms of their ability to predict the outcome of these
14 various experimental paths that we would be investigating.

15 (Slide.)

16 I would like to now turn to our proposed test
17 planning for future operational tests. I think I will try
18 to shorten this up a little bit, since basically the
19 methodology that we are proposing is very similar to what we
20 have just gone through in our loss of feed water transients.

21 Each experimental or each transient test series
22 would be defined by a single initiating event. We would use
23 the event tree approach, in that the testing would proceed
24 from a relative -- a baseline test where we looked at a
25 single initiating event, with all systems functioning as

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1 designed. We would increase the severity of each of the
2 tests by looking at multiple failure modes until we
3 eventually reached a limiting case which would indicate that
4 we would have -- that path would lead to core damage in
5 reactor.

6 Finally, the last point I would like to make is
7 that from these tests, we had hoped to define additional
8 testing that should be done in terms of looking at such
9 things as the potential for system recovery once the reactor
10 has stabilized in an off-normal condition, or possibly the
11 potential operator interaction, intervention which might
12 tend to mitigate the consequences once you were on a limited
13 case transient.

14 (Slide.)

15 The objectives then that we see for these tests
16 would be to look at the single initiating event, identify
17 the different paths that the reactor -- or in this case,
18 Semiscale could take as a result of multiple failure modes,
19 common mode failures.

20 We again would be looking at the capabilities and
21 limitations of the large reactor instrumentation in terms of
22 defining the particular state or condition of the reactor at
23 any point in the transient; and finally, we would be using
24 this information as a data base to evaluate codes in terms
25 of their capabilities to follow the transient and predict

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1 the outcome of the particular experiment that we were
2 looking at.

3 (Slide.)

4 There are a number of -- a wide variety actually
5 of operational transients which you could define to look at;
6 but generally, we feel you can categorize them into these
7 seven general categories. I won't dwell on these. I think
8 you can look at them for yourself.

9 I would like to point out that we have identified
10 in parentheses after each one of these transients,
11 short-term, medium- to long-term, and long-term.
12 Specifically what this is indicating is the time frame in
13 which -- within which we feel we could begin looking at this
14 particular category of transient.

15 The limitations are primarily on hardware. Tests
16 that we could run in the short term are those involving
17 minor system modifications. The medium- to long-term
18 transients would be those which involve some system
19 modifications such as the inclusion of a point-kinetics
20 model in our power control to simulate reactive feedback.
21 These modifications are of a less extensive nature and
22 therefore these -- those types of tests could be run on the
23 medium- to long-term.

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1 The only one at we have identified as a long-term
2 item, the anticipate transient without scram. These
3 particular transients for us to look at would require
4 inclusion of a reactive feedback model into our power
5 control area. We would have to consider secondary site
6 feedback effects. This would require a closed loop
7 secondary; and because of te relatively severe transients,
8 temperature transients that we would see, we would also have
9 to be looking at such things as the capabilities of our rods
10 to follow those transients and in fact survive through those
11 transients. It may require modifications in our heat rod
12 designs.

13 (Slide.)

14 In conclusion, the things we expect to get from
15 those operational transients are, first of all, we expect to
16 identify important thermal hydraulic behavior over a wide
17 range of transients, specifically covering those seven
18 generated categories that I presented on the previous slide.

19 We also would be looking at potential -- the
20 potential for operator interactions or operator actions
21 which might tend to mitigate the severity of the limiting
22 case transients. We would be looking at the ability of the
23 codes to predict these limited case transients and also to
24 define the operating margins that might be associated with
25 them.

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1 Finally, because we consider these tests to be a
2 scoping or paramethods nature, we would use these to help us
3 define future tests or future separate effects testing that
4 could be performed in Semiscale as well as by other
5 individuals or facilities.

6 I think that concludes my presentation. If there
7 are any comments --

8 DR. PLESSET: Could we see the event tree slide
9 for a moment?

10 MR. HARVEGO: Yes.

11 (Slide.)

12 DR. PLESSET: I want to look at the first three.

13 MR. HARVEGO: These three?

14 DR. PLESSET: Not the first one, the second one.
15 The TU one. TL you have a question. TU you have a no. Is
16 that the way you want it?

17 MR. HARVEGO: I looked at that. That's the way it
18 is defined in WASH 1400.

19 DR. PLESSET: What do you think?

20 MR. HARVEGO: This, to me, is saying that if we
21 have an auxiliary -- if we lose our normal feed water flow,
22 that we have auxiliary feed water; that you don't have to
23 open the pressurizer relief valve; that may be true for this
24 particular plant. I don't know whether it is true for all
25 plants. As a matter of fact, I doubt that it is true for

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1 all plants.

2 DR. PLESSET: I am not questioning the top one.

3 MR. HARVEGO: Okay.

4 DR. PLESSET: The next one.

5 MR. HARVEGO: This is saying that -- I think this
6 depends on the break size. You are not going to
7 overpressurize the system. The auxiliary feed water flow
8 may or may not keep the system -- allow the system to
9 depressurize.

10 This, to me, is saying that if you don't -- if you
11 do allow the system to depressurize, the HPIS will come on
12 and you get your core cooling.

13 DR. PLESSET: You are now assuming that it doesn't
14 come on?

15 MR. HARVEGO: If it doesn't come on -- and I am
16 not assuming this. This is what WASH 1400 says.

17 DR. PLESSET: You say there is no core damage?

18 MR. HARVEGO: There is no core damage.

19 DR. PLESSET: It is better if it does come on --
20 what are you saying?

21 MR. HARVEGO: What I am saying is this says it
22 will not. It means whether you depressurize or not.

23 MR. MATHIS: You don't need it. That's what it
24 says.

25 MR. HARVEGO: It says you don't need it. You

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1 could -- if it operates, fine; if it doesn't, it makes no
2 difference.

3 I think there are a lot of questions to these.

4 DR. PLESSET: Now, on the next one, let's look at
5 that. You get core damage you say?

6 MR. HARVEGO: Right.

7 DR. PLESSET: How do you explain that?

8 MR. HARVEGO: All I can say is that it would
9 appear that there isn't sufficient cooling with the HPIS for
10 that particular break size.

11 DR. PLESSET: It might or might not be the case?

12 MR. HARVEGO: It might or might not. This is
13 indicating that it is likely there would be core damage.
14 The extent, I don't know.

15 I think there is a lot of room for questioning
16 these event trees. That's one of the reasons why we have
17 taken this approach, to not only be looking at the potential
18 for variations in these events. These events assume that
19 either the system does or does not function as designed.

20 We would be looking at degrees. I think this
21 particular event tree was presented in WASH 1400. Whether
22 it is valid -- it is not valid for every plant. Whether it
23 is valid -- is truly valid for the particular plant that we
24 are using, I think that's up for questioning.

25 DR. PLESSET: Well, it juse seems a little gross,

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1 that's all, to me.

2 MR. HARVEGO: What seems gross?

3 DR. PLESSET: The conclusion you draw from this
4 particular event tree.

5 MR. HARVEGO: I am not drawing anything. That is
6 what is in WASH 1400.

7 DR. PLESSET: From that. That WASH 1400 is gross.

8 MR. HARVEGO: I think it is.

9 DR. PLESSET: Okay. Any questions of Mr. Harvego?

10 (No response.)

11 DR. PLESSET: I guess we are now ready to go to
12 Mr. Johnsen. He has been very patient.

13 MR. JOHNSEN: I am Gary Johnsen. I am with the
14 Semiscale program. Originally I intended to give you
15 gentlemen an overview of the results and significance of
16 results in the Semiscale program over the last year.

17 In the interests of time, however, I have a
18 abbreviated talk to focus on the Three Mile Island
19 simulations that were performed in Semiscale. Therefore,
20 there will not be a one-to-one correspondence between my
21 slides and what you are holding right now. In fact, it may
22 be very, very difficult to follow my presentation.

23 DR. PLESSET: We have trouble anyway.

24 MR. JOHNSEN: I just handed you a supplementary
25 list of slides, some of which will be in this, the rest of

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1 which will be in the original presentation handout.

2 In any event, as Dr. Wilkins pointed out
3 yesterday, and as has been mentioned previously, the
4 Semiscale program was fairly heavily involved with Three
5 Mile Island starting with the accident itself, during the
6 early crisis hours of the accident and for many weeks
7 following in terms of trying to gain more understanding of
8 what actually occurred.

9 (Slide.)

10 During the first few days of the accident, the
11 Semiscale program was asked to investigate particular ideas
12 with respect to recovering the system on the assumption --
13 under the educated assumption that a large noncondensable
14 gas bubble existed in the upper plenum, upper head region of
15 the TMI Plant.

16 We conducted two tests which we refer to as the
17 noncondensable bubble tests and the particular idea here was
18 to begin with the Semiscale system, having a condensable gas
19 bubble in the upper head region, scaled to what was then
20 believed to be the size of the bubble at TMI and then to
21 attempt to recover the system, simulate a recovery, that is,
22 in Semiscale by depressurizing the system to the point where
23 the residual heat removal systems could be made
24 operational.

25 Of course, the pressing question was could that be

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1 done without uncovering the core by virtue of movement of
2 the expansion of the gas bubble down into the core region.

3 DR. PLESSET: Did anyone here ever attempt to
4 check the bubble calculations, the size of the bubble?

5 MR. JOHNSEN: To my knowledge, we were given only
6 the results of those calculations by those who were on the
7 scene. However, I'm not privy to all of the bits and pieces
8 that filtered down to EG&G from that site, as far as what
9 went into the determining what the size of the bubble was.

10 Perhaps someone else in the audience could
11 elaborate on that.

12 MR. KAUFMAN: We ran the first test about four
13 hours after we were initially contacted. We began to modify
14 Semiscale to set up to run the tests. In that four hours,
15 we did not check the volume.

16 DR. PLESSET: I was just curious. I am just
17 trying to learn something.

18 MR. KAUFMAN: We did look at the SAR to conclude
19 that it was approximately the right upper volume; and it was
20 not off an order of magnitude, for example.

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1 DR. PLESSET: I think the calculations are suspect,
2 but I don't know for sure. I just thought you might have
3 happened to look at it.

4 DR. ROSZTOCZY: As you know, they have been
5 checked — there have been some check calculations done in
6 Bethesda that Sunday night. There were some calculations
7 which were later corrected. In the meantime, the bubble
8 disappeared.

9 DR. PLESSET: That's right. That's interesting.
10 I have been curious about it, have my own theory which I
11 won't expose now. (Laughter.)

12 DR. ROSZTOCZY: Maybe at some other place, some
13 other time?

14 DR. PLESSET: That's right.

15 Sorry to interrupt.

16 MR. JOHNSEN: That's all right.

17 In any event, those initial tests which I will
18 discuss very briefly were followed by eight tests which were
19 an attempt to, as faithfully as possible, simulate the actual
20 battery condition events, if you will, that occurred in Three
21 Mile Island as we understand them at that time on the
22 Semiscale system and to measure the response of the Semiscale
23 system; and by that process, to infer what actually occurred.
24 (Slide.)

25 I am getting ahead of myself to some extent.

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1 The objectives of those two different tests are
2 indicated here. Again the purpose of the noncondensable gas
3 bubble test was to plan, if we could -- to find out if
4 recovery by depressurization was a viable technique and the
5 purpose of the objective of the two-hour simulations, if you
6 will, starting from loss of feed water to core uncovering, was
7 to look at -- initially, any way -- to look at pressurizer
8 response in the hopes of ascertaining whether or not the Three
9 Mile Island pressurizer response was in fact a valid
10 indication of system liquid inventory as determined by looking
11 at the Semiscale results and also, of course, to investigate
12 what sort of problems we would have with Semiscale in terms
13 of conducting that sort of a simulation.

14 These objectives -- these latter objectives expanded
15 out as the tests evolved to look at more detailed behavior
16 vis-a-vis the initial pressure increase in the system when the
17 heat injected into the steam generators was lost -- whether or
18 not the code safety valves actuated. (Slide.)

19 Despite the fact that we were in somewhat of a
20 crisis environment early on, we nevertheless recognized that
21 we had in Semiscale a representation of a four-loop plant,
22 not a two-by-four plant. Consequently, we would have to look
23 at making changes in the system as we best could in the short
24 time available to better simulate the Three Mile Island B&W
25 plant.

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1 We made a number of changes. I don't want to dwell
2 too long on some of these. You can see that the first thing
3 we did was to take and remove the surge line that we had in
4 the plant, in the Semiscale system, replace it by one which
5 would be more representative of the B&W configuration in
6 terms of the dog leg which Tom Larson mentioned earlier, in
7 terms of the resistance, and also in terms of the resistance
8 of that line, as well as the elevation characteristics.

9 We also added in two separate lines exiting from the
10 pressurizer and terminating in orifices to simulate both the
11 code safety relief valves in the Three Mile Island plant as
12 well as the pile on the operated relief valve.

13 These -- especially the pile on the operated relief
14 valve orifices simulator was sized on the basis of the rated
15 flow capacity as stated in the SAR. Initially that told how
16 to design that orifice.

17 We also connected a line between the inlet annulus
18 in Semiscale and the upper plenum region to simulate the vent
19 valves in the B&W configuration. We found it necessary to
20 reduce the overall loop resistance so we could increase the
21 flow rate in the system on a scaled basis more closely
22 simulating that of Three Mile Island.

23 Also we added a orifice in the hot leg nozzle of the
24 intact loop to provide a more symmetric response in terms of
25 the draining characteristics of Semiscale. As you can picture

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1 in your mind, Semiscale has the two separate loops. Because
2 the intact loop represents three loops, it has a larger
3 diameter of pipe than the broken loop does.

4 However, the center lines are lined up. That
5 places the bottom of the intact loop cold leg pipe physically
6 lower than the respective location of the broken loop, so the
7 draining would be asymmetric otherwise.

8 This was a step that was particularly taken in
9 anticipation of the bubble tests rather than the overall
10 simulations that were later conducted. Despite these changes
11 to the system we realized we were stuck with some basic
12 limitations.

13 Tom Larson already discussed the subject of external
14 heat loss which in our system is disproportionately large. In
15 fact, in terms of our two-hour simulations of the Three Mile
16 Island accident, we utilized our heat loss, if you will, as
17 the primary heat rejection mechanism as opposed to simulating
18 the steam generator heat rejection in Three Mile Island.

19 The steam generator design was also much different.
20 We have U-2 steam generators versus the ones through in
21 Three Mile Island. This was mitigated by the fact we were not
22 relying on them as a heat rejection mechanism after the
23 initial completion of secondary liquid mass.

24 Also mentioned earlier by Tom was the primary
25 coolant pump degradation. We feel fairly confident in knowing

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1 or recognizing the fact that our pumps tend to degrade more
2 rapidly as a function of voiding in the system than would
3 the larger pumps in the B&W system.

4 Those were the primary limitations that we were
5 forced to just live with during these simulations.

6 I don't want to dwell very long on the results of
7 the bubble tests except to make two points. (Slide.)

8 The first is that in both of the tests we conducted
9 in terms of depressurizing the system with the gas bubble in
10 the upper plenum region, we found that in both cases, the
11 system could be depressurized in Semiscale again without
12 uncovering the core to the point where the residual heat
13 removal system could be activated.

14 Obviously this piece of information was not acted
15 upon in terms of recovering Three Mile Island, but still
16 provided useful information in the analysis which attended
17 recovery concepts.

18 Secondly, it is kind of interesting to note that
19 in the context of running those bubble expansion simulations,
20 if you will, that we learned that we could not operate both
21 Semiscale pumps simultaneously with the presence of that
22 noncondensable gas in the system. What we found was that
23 neither pump could generate a net positive suction head with
24 that degree of noncondensable gas in the system.

25 However, if we operated only one pump alone, it

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1 would force the gas into the other loop and then could — the
2 pump that was operating then could maintain coolability of the
3 core.

4 So that's an interesting result in terms of an
5 assessment of what sort of recovery techniques might be
6 suggested for the system; a PWR for example, had a large
7 amount of voiding in it, whether it was noncondensable gas
8 or steam.

9 I would now like to move on to some direct
10 comparisons. (Slide.)

11 I think this is probably the most interesting aspect
12 of these results. We have here a direct comparison between
13 the small Semiscale system and a big plant, between Semiscale
14 and Three Mile Island, in terms of some basic parameters that
15 were measured in both systems for the — what I call the
16 two-hour simulations.

17 Recognize that when these simulations were conducted
18 there were several unknowns and still are some unknowns in
19 terms of what was the — what were the primary boundary
20 conditions on the Three Mile Island plant. Principally here
21 we are talking about HPIS operation as a function of time,
22 as well as makeup and let-down flow rates as a function of
23 time.

24 These are not well established at this time and
25 certainly to have a influence on the overall mass and energy

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1 balance in the system.

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2 We took the information — the best information we
3 had from the Nuclear Regulatory Commission and factored that
4 into a test plan for these two-hour simulations. Here we
5 are looking at a comparison of the system pressure as a
6 function of time for approximately a two-hour period. As you
7 can see, the agreement is generally pretty good between the
8 Semiscale system and Three Mile Island. I think this is
9 primarily attributable to the fact that we were able to
10 simulate the mass discharge through the pile on the operated
11 relief valve fairly faithfully with our orifice. (Slide.)

12 Here is a comparison of hot leg temperature in the
13 two systems. The agreement is qualitatively fairly good; and
14 notice especially that the time at which superheating is noted
15 in the Three Mile Island plant was fairly closely simulated
16 by the Semiscale system, following the termination of pump
17 power. (Slide.)

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1 Here in this slide we are looking strictly at
2 Semiscale data. This sort of summarizes the tail end of
3 our simulations.

4 Here we are looking at the last few moments of the
5 transient, if you will, culminating with core uncover. We
6 are looking at the pressurizer level in Semiscale, in this
7 case multiplied by a factor of four to conform to the eleva-
8 tion in Three Mile Island.

9 The middle plot is a -- is the collapsed liquid
10 level in the Semiscale core; and the bottom line is the mid
11 core temperature measurement in Semiscale.

12 Now, you will notice that after the pumps were
13 shut off in Semiscale, we see a termination of the oscillatory
14 behavior in the collapsed liquid level in the core. It evens
15 out and remains level for quite some time.

16 During this period of time, what occurred was a
17 draining of liquid from the upper portions of the system; the
18 steam generators into the hot legs into the core to balance
19 the boil off that was occurring in the core so that the level
20 stays fairly stable.

21 When this draining is complete, we then see a
22 boil off, a fairly straight boil off of the core. We see
23 the collapsed liquid level dropping. When the collapsed
24 liquid level falls low enough in the core to uncover, the
25 location associated with this thermocouple, you see the

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1 temperature take off. It is significant, I think, to note
2 that during all of this voiding of the system in the core
3 region, you see the pressurizer remains full.

4 The discussion of the validity of these results
5 with respect to typicality of Semiscale and TMI was discussed
6 earlier, so I won't touch upon that.

7 (Slide.)

8 It's interesting to look at what was going on in
9 the core region of Semiscale during the time immediately
10 adjacent to the termination of pump operation.

11 Because we have in Semiscale the external downcomer,
12 we can make density measurements directly into the core. In
13 this plot, what we are looking at is the density measurement
14 at three different axial locations in the core. The lower
15 part of the core shows the least amount of void, obviously.
16 As we progress up the core, we see higher voiding.

17 Note that this -- this period incidently is the same
18 period we showed a minute ago where the liquid level was
19 fairly stable in the core.

20 Note that during the boil off, which occurs here,
21 what we see is a progressive drop off in the density starting
22 from the upper part of the core, where we see a rapid turn-
23 over in this density, indicating that fluid has passed by that
24 axial measurement, and progresses up the core.

25 Now, this sort of a measurement enabled us in

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1 conjunction with the collapsed liquid level measurement to
2 infer the relationship between a collapsed level and a froth
3 level.

4 (Slide.)

5 That is shown on this next slide here. In other
6 words, by noting the times at which the density takes the
7 sharp drop at the various elevations, you can infer where the
8 froth level was as a function of time. That's shown here.
9 We can then compare that with the collapsed liquid level as
10 shown here as measured by the delta p measurement of the core.

11 We not very interestingly that these two levels
12 tend to come together. This is not totally surprising since as
13 the boil off occurs, what we have is less and less net energy
14 entering the fluid, and, therefore, less violent boiling as
15 the boil off proceeds down the core.

16 So when we reach an elevation that's approximately
17 slightly below mid core, we see that we have practically
18 collapsed all of the froth, and we have perhaps a very small
19 froth region. This has implications in terms of the coolabil-
20 ity of those exposed core rods above that region.

21 We were able to look at the heat up data from the
22 Semiscale core rods, and by performing inverse calculations,
23 compute the heat flux at the various elevations above the
24 collapsed level.

25 This plot illustrates the computed heat transfer

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1 coefficient at one particular axial elevation as a function
2 of its position above the collapsed liquid level.

3 Again, we could construct a number of such curves
4 for each -- for each elevation to have its own curve.

5 This shows that as the collapsed liquid level
6 drops down in relation to the position of this particular
7 location of the core, that the heat transfer coefficient drops
8 down.

9 Now, in order to calculate this heat transfer
10 coefficient, we required the use of an assumed bulk fluid
11 temperature at the same axial elevation. In this case,
12 what we have done is assume saturated vapor. You will recog-
13 nize the fact that the vapor will not be saturated, the fact
14 it will superheat as that vapor rises through the core in the
15 voided region. However, in the case of Semiscale, we are --
16 our measurements are insufficient in terms of measuring vapor
17 temperature because of radiation.

18 In any event, we could take this heat transfer
19 information and together with other data from Three Mile Island,
20 namely data relative to the power and axial power shape and so
21 on, together with liquid level data as inferred from the ex-
22 core detectors in Three Mile Island and construct, if you will,
23 a liquid level as a function of time in Three Mile Island and
24 which, taken together with the heat transfer coefficient
25 measurement or calculations, I should say, from Semiscale, would

1 yield a prediction for the heat transfer coefficient as a
2 function of elevation in Three Mile Island.

3 That's illustrated on this slide here.

4 (Slide.)

5 We are looking at the period essentially from
6 the time the pumps were -- the last pump was shut off in
7 Three Mile Island to the time when the pump was reactivated.
8 These are the predictions, again, based on that methodology
9 I have just described for the heat transfer coefficients at
10 various elevations in the Three Mile Island core.

11 When these results were applied to a conduction
12 calculation in Three Mile Island, which took into account the
13 exothermic zirc-water reaction on the external portions of
14 the rods, it yielded a temperature prediction that you are
15 looking at right now.

16 (Slide.)

17 This showed or suggested that below about the 2.9
18 meter elevation in Three Mile Island, that the core remained
19 cool, but that above 2.9 meters, the temperatures reached
20 elevated levels. Depending upon which side of the uncertainty
21 band and the heat transfer coefficient that we computed you
22 wished to use, you would either obtain temperatures that
23 essentially went out of sight or leveled off at nonetheless
24 very high temperatures where, in fact, the zirc-water reaction
25 would be very important, and also where core damage could

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1 obviously result.

2 (Slide.)

3 Summarizing, then, in terms of what we think is
4 significant from these tests, first of all we think it's
5 rather significant that the overall hydraulic trends demon-
6 started in Three Mile Island were fairly faithfully reproduced
7 in our system despite the fact that we were not designed to
8 simulate that system.

9 I think that because this accident is predicated
10 on mass and energy balances primarily, that we were able to
11 achieve a fairly good simulation.

12 Secondly, as evidenced by the results I showed you
13 earlier, it's fairly clear now -- and I don't think there
14 is any argument on anyone's part -- that the pressurizer
15 from Three Mile Island -- liquid level in the pressurizer was
16 not a valid indication of system liquid inventory.

17 Thirdly, that the whole area of core uncover heat
18 transfer deserves a great deal of additional attention.
19 Dr. Rosztoczy suggested this earlier, and we certainly agree.
20 In fact, the heat transfer that we saw in Semiscale during the
21 core uncover period is very, very poor and gets worse as the
22 level -- liquid level in the core drops.

23 This is an area that I think is particularly
24 relevant in terms of small break calculations.

25 Again, the fourth point is really an indication of

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1 what I showed you a minute ago, that our results when
2 projected on the basis of what I have described it to Three
3 Mile Island and suggest significant core damage could have
4 resulted. Again, it's -- there's probably no argument there.

5 Lastly, these tests were useful in terms of defining
6 those particular areas in Semiscale that would require atten-
7 tion in the future for looking at very long transients;
8 specifically what I mentioned earlier, the heat loss problem;
9 pump leakage was another area Tommy Larson talked about earlier
10 which became a factor in these tests.

11 That concludes my presentation.

12 DR. PLESSET: I guess there are no questions.

13 Thank you very much. We appreciate that.

14 Well, we are near adjourment time. I was going to
15 see if our consultants wanted to make general remarks, or
16 specific remarks, before we adjourn.

17 (No response.)

18 DR. PLESSET: I guess not.

19 MR. LIPINSKI: I have a general comment.

20 DR. PLESSET: Yes, please.

21 MR. LIPINSKI: In Dr. Rosztoczy's view graphs,
22 he indicated there was a desire to have the reactors run
23 some mild transients experimentally; is that correct?

24 DR. ROSZTOCZY: Yes, that's correct. I am not sure
25 I characterized it as mild.

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1 MR. LIPINSKI: In the view graph prior to that, it
2 addressed the LOCA events. I forget what the magic words were.
3 I think confirmatory or something like that.

4 I don't think you specifically said you wanted to
5 do any experimental work on small LOCA's on an actual plant;
6 am I correct?

7 DR. ROSZTOCZY: Not on the last point. We have
8 requested that they do experimental work on the small loss
9 of coolant; but we have asked that at least the one test
10 showing the normal behavior of the pressurizers as designed.
11 So the pressurized sufficiently that the high pressure safety
12 injection, the safety injection and the low pressure injection
13 comes into play, a demonstration test for that.

14 We asked for a typical test for the case when the
15 break is smaller than the size and the pressure hangs up on --
16 at the pressure level which is usually close to the second
17 area system pressure, and then show the recovery from this
18 condition.

19 Then we asked for a third test which is even smaller
20 break size than this, where the pressure after an initial
21 drop is actually going to increase and repressurize to some
22 higher level and the show how it should recover from that test.

23 MR. LIPINSKI: Is this per plant or per typical
24 plant?

25 DR. ROSZTOCZY: Just one. One for each on the

rl-9
DL
1 Semiscale system.

2 Then there were some additional requests. We wished
3 to demonstrate some of the basic phenomena which is presently
4 being predicted to take place in some of the systems. We
5 talked about nature circulation in the ones through steam
6 generator. We were wondering if it were possible to set up some
7 tests that may be for a given time period you can stay in a
8 steady condition in this type of heat transfer demonstrate
9 how it works, demonstrate how you get into it, and what might
10 possibly interrupt this.

11 MR. LIPINSKI: Are you talking about tests or tests
12 on a natural system?

13 DR. ROSZTOCZY: No. All of these tests are for
14 Semiscale.

15 MR. LIPINSKI: I am talking about tests on a full
16 size reactor. On transients you specifically said these tests
17 would be performed on an actual reactor system, a full sized
18 reactor to get confirmatory results.

19 I was going to pose a question as to whether you are
20 looking for equivalent information on small LOCA's on a full
21 sized reactor system.

22 DR. ROSZTOCZY: I am sorry. I misunderstood the
23 question. For all the LOCA type of tests, we are not pro-
24 posing any LOCA type of test to be done on a full sized
25 reactor. We are sponsoring all of those -- for transients

1 where normal limit is that you do not go into DNB. We are
2 proposing carefully selected typical tests on actual reactors.

3 MR. LIPINSKI: In view of the questions on
4 confirming codes, experiments, comparison to experiments,
5 three dimensional systems, multiple loops, is there any bene-
6 fit to consider full sized tests where the risk is relatively
7 small?

8 DR. ROSZTOCZY: A good example of that would be the
9 General Electric tests. They had a code developed 20 years
10 ago that has been used for the safety evaluation in the design
11 of the boiling water reactors. Basically, the code simulated
12 the transient and calculated the change in the critical power
13 ratio during the transient.

14 The plants then are designed to have a sufficient
15 margin in the operating condition that they will be removed
16 far enough from the point when you might go into a hot point DNB
17 to cover this critical change during the transient.

18 It is an important part of the design of the plant.
19 This code has been used to predict the tests. As a matter of
20 fact, it has been used for more than that. It has been used
21 for the safety evaluation of the reactor for those tests before
22 we could permit those tests in an actual reactor. We wanted
23 an assurance that the concept of the tests are acceptable.

24 There is calculation based on that that permitted
25 them to run the tests, They ran the tests, The data CPR changed

1 for the three-date appointment they had planned. It was 100
2 percent up to either 300 percent. At the same time, the margin
3 that was in that calculation was on the order of maybe 20
4 percent. So we found a huge difference between the code
5 calculation which had been performed continuously for 10
6 years or so, the licensing of these plants, and within the
7 physical behavior that one could observe an actual reactor
8 which had all the feedback.

9 After obtaining this, they went into the detail to
10 analyse where the difference was coming from. It has been
11 tracked down to the time response of the void collapse.
12 The void apparently collapsed in the actual reactor faster
13 than in the calculation which provided the reactor with the
14 feedback, and the activity feedback came somewhat earlier in
15 time. There was no other position for that.

16 That resulted in a higher power excursion for a
17 shorter period of time.

18 This is an example of a test performed on an actual
19 reactor that can point out deficiencies that existed for a
20 number of years, but there was no way to check on it. The
21 code response was the type of response that you would expect.
22 There was nothing in it that would say this is unrealistic.
23 It was only delayed somewhat in time, and that caused the
24 problem.

25 When the PWR's -- they can't -- point to a certain

rl-12
DD

1 to a certain problem of here is something that might be wrong.
2 Nevertheless, we would like to see some challenging test
3 which goes through on a reasonably trusted transient, that it
4 challenges the code. Obviously, it has to be within the
5 safety limits of the reactor and see if it turns out the same
6 way as the calculations.

7 The PWR's the dynamics of the steam generator
8 stop that from performing these calculations.

9 They are usually evaluated for planned operations.
10 There is no data for anything more drastic than that.

11 We feel it would be beneficial to have some
12 selected tests in probably one reactor, or two typical ones
13 and compare those against the calculations.

14 MR. LIPINSKI: No further comment.

15 DR. PLESSET: Any other questions?

16 (No response.)

17 DR. PLESSET: Let me ask a question, then, of Zoltan.

18 What you are planning is a series of tests in
19 Semiscale in which you have a version which is B and W, a
20 version which is CE, one which is Westinghouse; is that
21 correct?

22 DR. ROSZTOCZY: We are doing calculations for each
23 of those.

24 DR. PLESSET: Calculations? I thought you were
25 going to have tests?

rip 13
DL

1 DR. ROSZTOCZY: The tests, I believe, are being
2 set up to simulate some of the basic phenomena expected in
3 each of these reactors. I think the CE and the Westinghouse
4 are the same, so I don't think we are talking about any
5 different tests there.

6 DR. PLESSET: So the machine would be the same?

7 DR. ROSZTOCZY: I believe it would basically be the
8 same.

9 For the B and W case, there would be some differences
10 because of the ones through steam generator that introduces
11 a kind of difficulty.

12 DR. PLESSET: My question, then, is somewhat
13 reduced. Would you feel that it was significant to make the
14 changes in the machine, Semiscale, for these changes relative
15 to the plants? Would that be lost in the errors in the
16 similitude which Semiscale has, of necessity?

17 In other words, they are all pressurized water
18 reactors. That's the zero order approximation.

19 Now then, you get into a higher order. One is
20 four loop. Westinghouse. U-2. One is a two loop. The other
21 is once through. With all of these refinements -- will these
22 refinements be different enough to justify changing the
23 machines before you do the experiments? Do you see what I am
24 trying to get at?

25 DR. ROSZTOCZY: I think so. Let me try to explain

rl- 14
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it through an example about the circulation problem.

1tDD

1 In the B&W design, the ones through
2 steam generator, through analysis, through thinking and
3 doing certain calculations, both B&W and us, we kind of
4 arrived at a point where both sides agree. We found two
5 ways of how the natural circulation can be interrupted in
6 those systems. One of them is the formation of a bubble in
7 the top of the candy cane which can prevent natural
8 circulation. The second one is a condition when the water
9 in the secondary side of the steam generator is at the low
10 level but if there is no spraying of any more feed water,
11 for example, because you can't control to that level, could
12 prevent natural circulation in the loop.

13 The second type of actual -- stopping of the
14 natural circulation we believe actually happened at TMI.

15 . To demonstrate these two, you need a once-through
16 steam generator in the loop. You would have to take out the
17 steam generator which is there now and put in a different
18 one.

19 At the same time through our thinking and this
20 calculation, we could not postulate any other than this.
21 The test might bring attention to something else. So we
22 think it's important.

23 Now, when you go with the same question, the
24 question of natural circulation to a U-2 type of steam
25 generator, both manufacturers who have U-2 type of steam

ltDD
1 generators describe a quite different process. They are
2 saying the basic generator, even if there is just a small
3 amount of water on the secondary side, then the steam has to
4 go up on a small cooled section on the steam generator
5 tubes.

6 Some of the steam will condense there. Water is
7 going to drop down against the steam flow and then leak back
8 to the top of the core and provide core cooling.

9 So they believe that actual stopping of the
10 natural circulation will not happen in the steam generator
11 unless they dry out the steam generator.

12 They have calculations to support this, but we
13 have not seen any physical demonstration that it really
14 works this way; and there will be enough water dropping back
15 provided you have so much water in the secondary side of the
16 steam generator. We would like to see a demonstration of
17 this type of phenomena.

18 Whether I have 8,000 tubes in the steam generator,
19 which the normal PWR has, or whether we just have a handful
20 of tubes there, that's somewhat different. The phenomena, I
21 think, can be demonstrated. We are asking for the
22 demonstration. This alone would warrant the insertion of a
23 different steam generator into the loop for one type of test
24 as opposed to the second type of test.

25 DR. PLESSET: I see your concern. What I was

1tDD 1 trying to ask about are the atypicalities of Semiscale large
2 enough so that you might not trust the results either way?

3 DR. ROSZTOCZY: I don't think so.

4 MR. ALLEMAN: Does it scale?

5 DR. PLESSET: That's the question. Does it scale?
6 If it doesn't, what do you need to do?

7 DR. ROSZTOCZY: I used on my slide the word
8 demonstration test. Demonstration is meant along these
9 lines, that we intend to demonstrate through these tests
10 that this type of actual circulation does work, does work in
11 a system with a few tubes.

12 If we can demonstrate that, then I think we are in
13 a much better position to pass a judgment as to how
14 effective this phenomena is in the large system. If we have
15 never seen any demonstration of this, we are in a much
16 bigger position.

17 DR. CATTON: Wouldn't you be better off to run
18 this particular test as a separate effects test? The steam
19 generator U-2 refluxing?

20 DR. ROSZTOCZY: I am not sure what is the
21 terminology that has been used here. You can think of it as
22 a separate effects test. Part of our thinking was one of
23 these can be run as more than a series. These are not
24 tests -- if it works properly, this is not a test which has
25 to be terminated. Some of the blowdown you have to

5468 28 04

1tDD

1 terminate.

2 You run for a steady state and you see how this
3 sets in. Then from this, you might transfer into a
4 different mode of heat removal and you demonstrate that for
5 a half an hour and transfer into a third mode and
6 demonstrate that for a half hour.

7 DR. CATTON: This would not necessarily be in
8 Semiscale?

9 DR. ROSZTOCZY: I am not as familiar with the
10 details of how difficult it is to set up some separately —
11 so-called separate tests as opposed to taking this equipment
12 into Semiscale and running it in Semiscale when you have all
13 the instrumentation all ready.

14 I don't know which is cheaper. I don't know
15 which is easier to do.

16 Theoretically, yes, some of this could be run as
17 somewhat of a separate test. It would require a full loop
18 in the heated core.

19 DR. TONG: I would like to make a comment. In
20 separate effects, you need another facility. If you wanted
21 a high pressure, you have to have a separate facility. You
22 fabricate a high pressure group to test the separate
23 effect. So Semiscale, you can interpret it as a separate
24 effect in the components.

25 If you have any thought about loop circulation

6468 28 05

1tDD

1 time, then you can -- either you can incorporate it by
2 calculation, say this data should be modified by -- divided
3 by, or you get all this circulation.

4 You do find out that a separate effect in the
5 steam generator exists.

6 You can treat Semiscale --

7 DR. CATTON: You make the rest of the Semiscale
8 just your high-pressure system then?

9 DR. TONG: You can do that. I really think that
10 it's not quite economical or practical. To do it, many
11 separate effects at high pressures would have to be done.

12 DR. CATTON: It's a matter of the view you take
13 when you set up a test at the outset?

14 DR. PLESSET: Well, clearly we don't have enough
15 in the way of experimental facilities that work at high
16 pressure.

17 DR. TONG: Right.

18 DR. PLESSET: Do you want to make any final
19 remarks, Dr. Tong?

20 DR. TONG: Well, I think that the last statement
21 is almost the same. The time schedule is so tight. We wish
22 we could use all the existing facilities with modifications
23 and satisfy the licensee's need; and also all the other
24 people's need.

25 But in doing this, I urge my contractors to

ltDD 1 emphasize their analytical effort. I don't mean cold
2 calculations; analytical analysis. Use the hardware, the
3 facilities to analyze or to understand the behavior of the
4 facility, what would come out, which is typical, which is
5 not typical, and get advanced information before the test
6 results come out and avoid a misunderstanding. That's all I
7 wanted to say.

8 DR. PLESSET: Thank you.

9 Mr. Kaufman, do you want to have a last word?

10 MR. KAUFMAN: Very briefly. I would like to come
11 back to the opening remarks that I made. We believe that
12 our product here is to develop understanding of the
13 processes and the system behavior that characterize
14 postulated accident conditions. We think that the LOFT
15 program and the Semiscale program and indeed the code
16 development efforts focus on the areas that are of the
17 greatest concerns now.

18 We believe that the programs we put together are
19 responsive to the requests of NRC and are responsible in
20 terms of approaching the important issues.

21 I think any one part of our program without the
22 other would put us in a somewhat weaker position.
23 Specifically, to conduct meaningful tests in LOFT, it
24 implies that we have sufficient understanding of the
25 phenomena that we can design the tests and locate the

1 instruments and prepare our operators for what they will
 2 experience. To do that well, I think we need the tests and
 3 facilities like Semiscale. Whether one calls those separate
 4 effects or integral, I think that is somewhat moot.

5 The point is issues are identified, phenomena are
 6 explored; and that adds to and contributes to our
 7 understanding.

8 I could make the same remarks for the codes and
 9 code development because again that's another adjunct to
 10 characterizing the degree of our understanding.

11 I think that's the overall program that we have
 12 going on, and we appreciate very much your comments and
 13 observations.

428

14 DR. PLESSET: Well, thank you. I think we have
 15 had a good meeting. We look forward to the next one.

16 With that, we will adjourn.

17 (Whereupon, at 2:40 p.m., the meeting was
 18 adjourned.)

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Larson

8/28/79

SEMISCALE SCALING

T. K. LARSON
ACRS MEETING
AUGUST 27, 1979



1004 126

SEMISCALE SCALING PHILOSOPHY

- APPROACH
- CRITERIA
- COMPROMISES

DEFINITION OF SCALING

THE TECHNIQUE BY WHICH A FACILITY IS CONSTRUCTED TO
ALLOW THE IMPORTANT PHENOMENA IN A REFERENCE SYSTEM
TO BE REPRESENTED IN A GEOMETRICALLY SMALLER SYSTEM

REDUCED SCALE EXPERIMENTS

- IT IS IMPOSSIBLE TO MAINTAIN GEOMETRIC, KINEMATIC, AND DYNAMIC SIMILARITY AT EVERY POINT IN THE EXPERIMENT FOR ALL TIME PERIODS OF CONCERN
- DISTORTIONS WILL EXIST IN THE EXPERIMENT RELATIVE TO THE LARGE SYSTEM FOR PHENOMENA WHERE MULTIDIMENSIONAL EFFECTS ARE IMPORTANT

POSSIBLE SCALING APPROACHES

- LINEAR
- DIMENSIONLESS NUMBERS
- VOLUME

MOD-3 SYSTEM SCALING

- VOLUME SCALING APPROACH
- WESTINGHOUSE TROJAN BASE PLANT FOR LOOP SCALING
- WESTINGHOUSE UHI BASE PLANT FOR VESSEL
- INITIAL SCALING DONE BY EG&G
- NRC, WESTINGHOUSE AND CONSULTANT REVIEW
- FINAL SCALING WITH NRC AND REVIEW GROUP

REFERENCE PWR AND MOD-3 VOLUME DISTRIBUTION

COMPONENT	REFERENCE PWR		MOD-3	
	VOLUME (M ³)	% TOTAL	VOLUME (M ³)	% TOTAL
DOWNCOMER	31.37	9.75	0.0222	10.66
UPPERHEAD	22.53	7.0	0.0140	6.75
UPPER PLENUM	17.49	5.44	0.0112	5.38
CORE	18.31	5.69	0.0106	5.08
LOWER PLENUM	27.52	8.56	0.0157	7.54
GUIDE TUBE	8.45	2.63	0.0015	0.74
SUPPORT TUBE	1.19	0.37	0.0004	0.19
TOTAL VESSEL	126.84	39.45	0.0756	36.34
INTACT LOOP	153.62	47.79	0.1037	49.86
BROKEN LOOP	41.00	12.76	0.0287	13.79

131 A001

1004 132

IMPLICATIONS OF VOLUME SCALING IN MOD-3

- MAINTAIN GEOMETRIC, DYNAMIC, AND KINEMATIC SIMILARITY IN THE CORE AND STEAM GENERATOR
- PIPING LENGTHS NOT MAINTAINED
- PIPING DIAMETER NOT MAINTAINED
- PIPING SURFACE AREA NOT MAINTAINED
- L/D OF SOME COMPONENTS IS LARGE

MOD-3 COMPONENTS

- DESIGNED WITH UHI IN MIND
- TRIED TO MINIMIZE DISTORTIONS
- HEAVILY INSTRUMENTED TO HELP QUANTIFY DISTORTIONS

MOD-3 CORE

- SCALING CRITERIA - VOLUME, LENGTH, POWER, AXIAL POWER PROFILE
- COMPROMISES - ELECTRICAL ROD, SPACER GRIDS, STRUCTURE SURFACE AREA
- TREATMENT

DESIGN

COMPUTERIZED POWER CONTROL

INSULATION

EXPERIMENTAL

POWER CONTROL TESTING

SERIES 7 RESULTS

ANALYTICAL

CORE POWER DETERMINATION

CONDUCTION CALCULATIONS

1004 135

1004 135

SEMISCALE MOD-3
INTACT AND BROKEN LOOP

SCALING CRITERIA:

HYDRAULIC RESISTANCE, VOLUME, PUMP SUCTION LEG DEPTH

SCALING COMPROMISES:

FLOW AREA, LENGTH, SURFACE AREA, INSTRUMENTATION

TREATMENT:

EXPERIMENTAL:

HYDRAULIC RESISTANCE TESTS

ANALYTICAL:

CONDUCTION ANALYSIS

SEMISCALE MOD-3

STEAM GENERATORS

SCALING CRITERIA:

VOLUME, SURFACE AREA, PRESSURE DROP, LENGTH

SCALING COMPROMISES:

INTACT LOOP (SCALED FROM LOFT)

BROKEN LOOP SECONDARY SIDE VOLUME

TREATMENT:

EXPERIMENTAL:

INVESTIGATION OF INFLUENCE OF BROKEN LOOP

STEAM GENERATOR (TEST SERIES 7)

EVALUATION OF INTACT LOOP STEAM GENERATOR

HEAT TRANSFER (MOD-1 TEST SERIES 1)

ANALYTICAL:

RELAP CALCULATIONS

881 4001

1004 137

SCALING CONSIDERATIONS FOR LARGE BREAKS

POSITIVE POINTS:

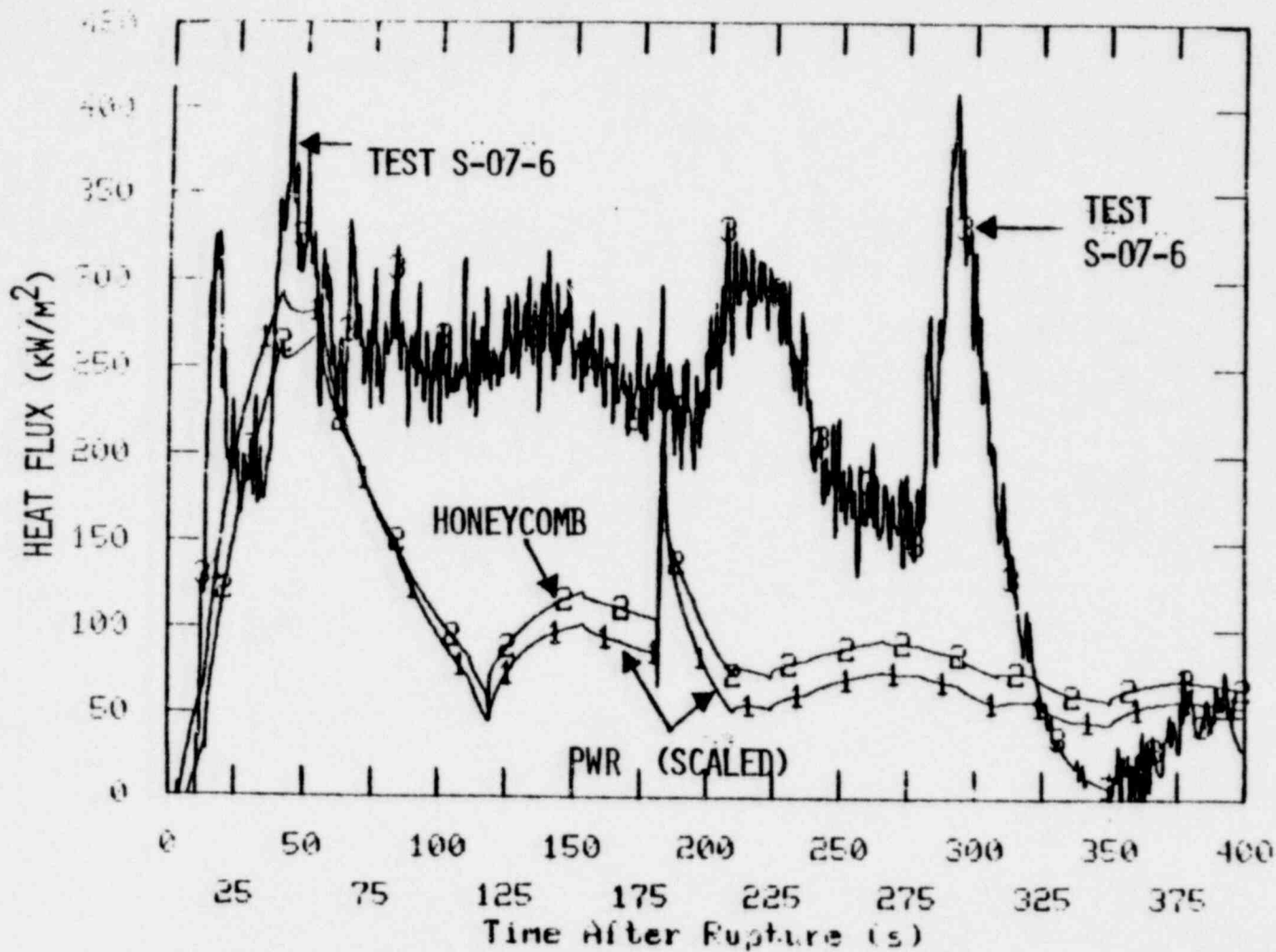
- RESISTANCE DISTRIBUTION IS CORRECT
- ENERGY AND MASS DISTRIBUTION IS CORRECT
- CORE GEOMETRY
- TIME SCALE
- RELATIVE ELEVATIONS MAINTAINED

CONCERNS:

- METAL STORED ENERGY AND SURFACE AREA TO FLUID VOLUME RATIO
- LOOP VELOCITY
- PUMP CHARACTERISTICS
- STEAM GENERATOR CHARACTERISTICS

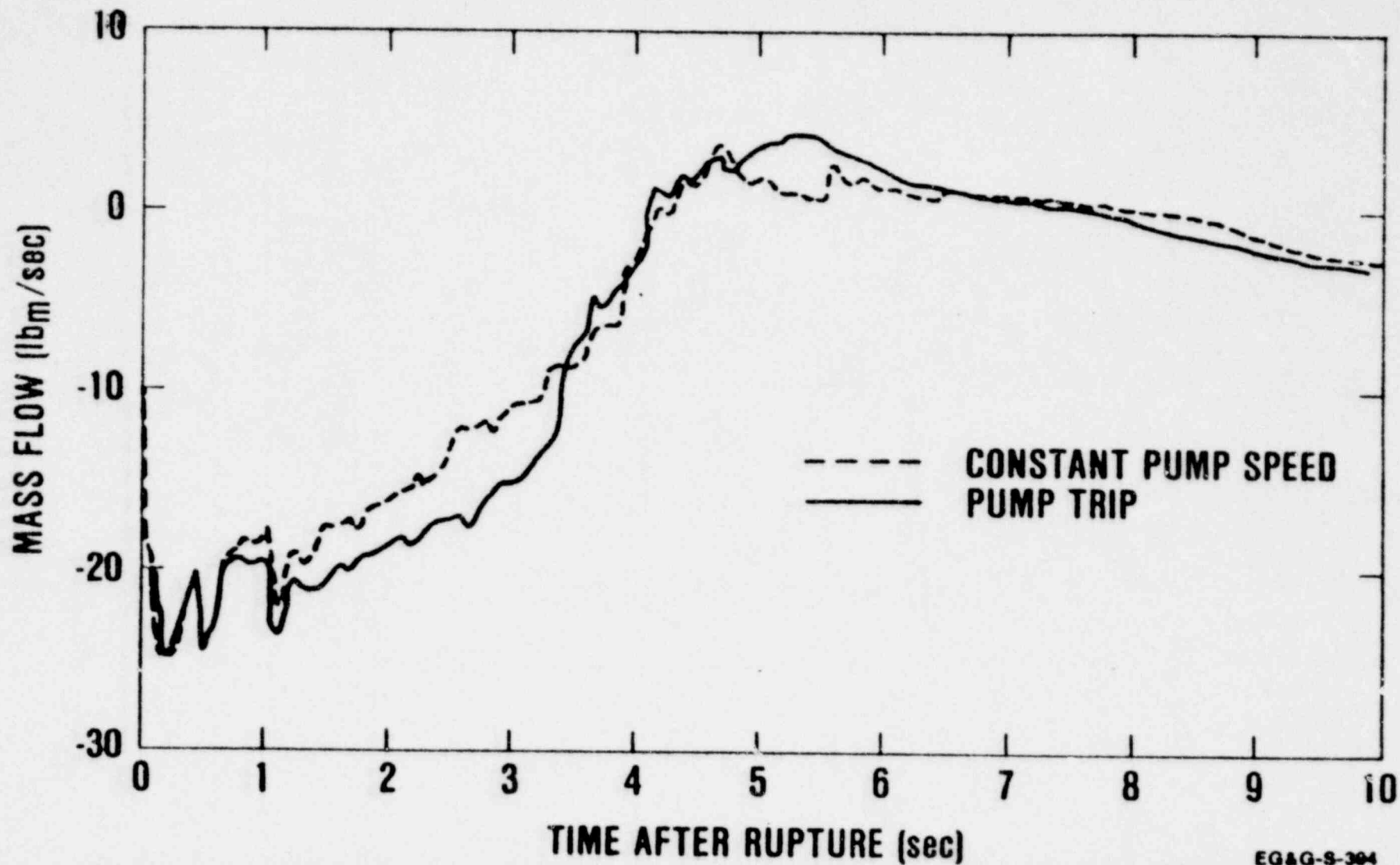
POOR ORIGINAL

PROPERLY SCALED HEAT FLUXES FOR TEST S-07-6 AND CONDUCTION LIMITED PWR AND HONEYCOMB DOWNCOMER



1004 139

DOWNCOMER MASS FLOW

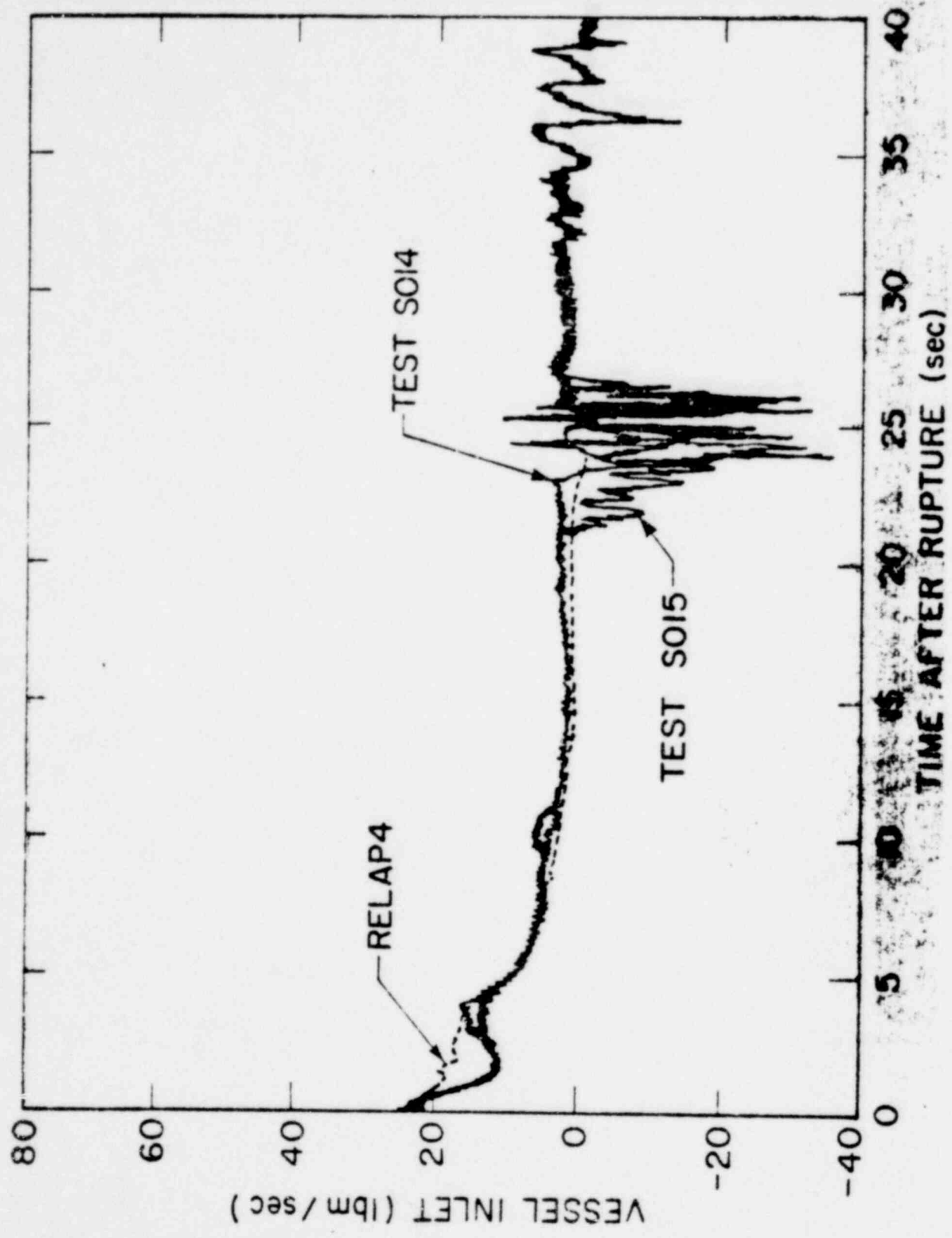


EG&G-S-394

1004 120

1004 140

FLOW RATE AT INTACT LOOP COLD LEG



FLOW RATE AT INTACT LOOP COLD LEG

VESSEL INLET (lbm/sec)

TIME AFTER RUPTURE (sec)

1004 145

141 1004

POOR ORIGINAL

SCALING CONSIDERATIONS FOR SMALL BREAKS

POSITIVE POINTS:

- LOOP SEAL ELEVATIONS CORRECT
- ACTIVE BROKEN LOOP COMPONENTS
- CORE LENGTH AND GEOMETRY
- BREAK AREA EASILY CHANGED

CONCERNS:

- HEAT TRANSFER (LOSSES, PIPING THERMAL TIME CONSTANT)
- FLOW REGIMES / VELOCITIES
- PUMP BEHAVIOR
- CRITICAL FLOW
- STEAM GENERATORS
- DIMENSIONALITY

ESTIMATED COMPONENT HEAT LOSSES IN SEMISCALE

COMPONENT	PRESENT HEAT LOSS (kW)	RESOLUTION	ESTIMATED MINIMUM HEAT LOSS (kW) *
VESSEL	40	HONEYCOMB, EXTERNAL HEATING, QUALITY INSULATION	5
DOWNCOMER	13		5
LOOP PIPING			
BROKEN	18	HEAT TAPES QUALITY INSULATION	3
INTACT	25		3
INSTRUMENTATION	13	MODIFICATION IN APPROACH	7
MISCELLANEOUS	<u>8</u> 117	ATTENTION TO QUALITY OF INSTALLATION OF INSULATION	<u>3</u> 26

* WITH THE EXTENSIVE USE OF HEAT TAPES

AA1 1001

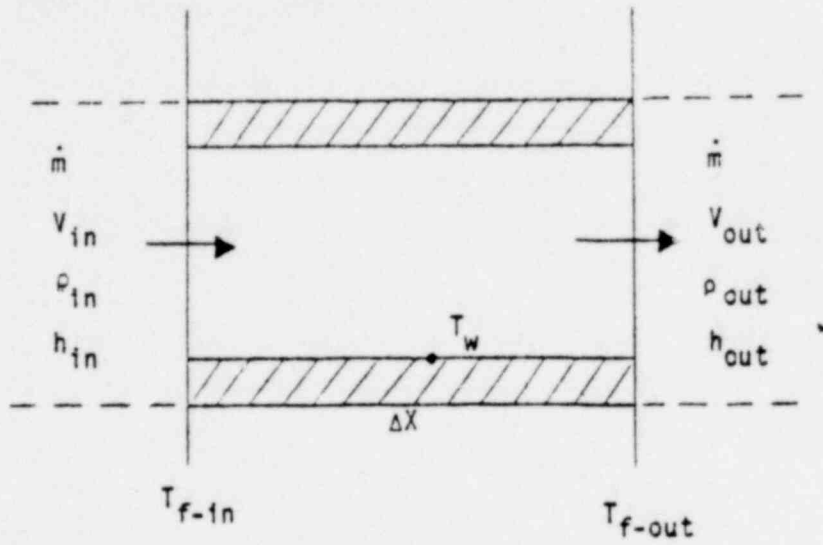
1004 143

CP1 4001

EXTERNAL HEAT LOSSES COMPARISON TO LOFT, PWR, SEMISCALE

	PWR	LOFT	SEMISCALE
TOTAL LOSS (kW)	1500 - 2000	257	120
% OF CORE POWER	0.07	0.5	6.

1004 144

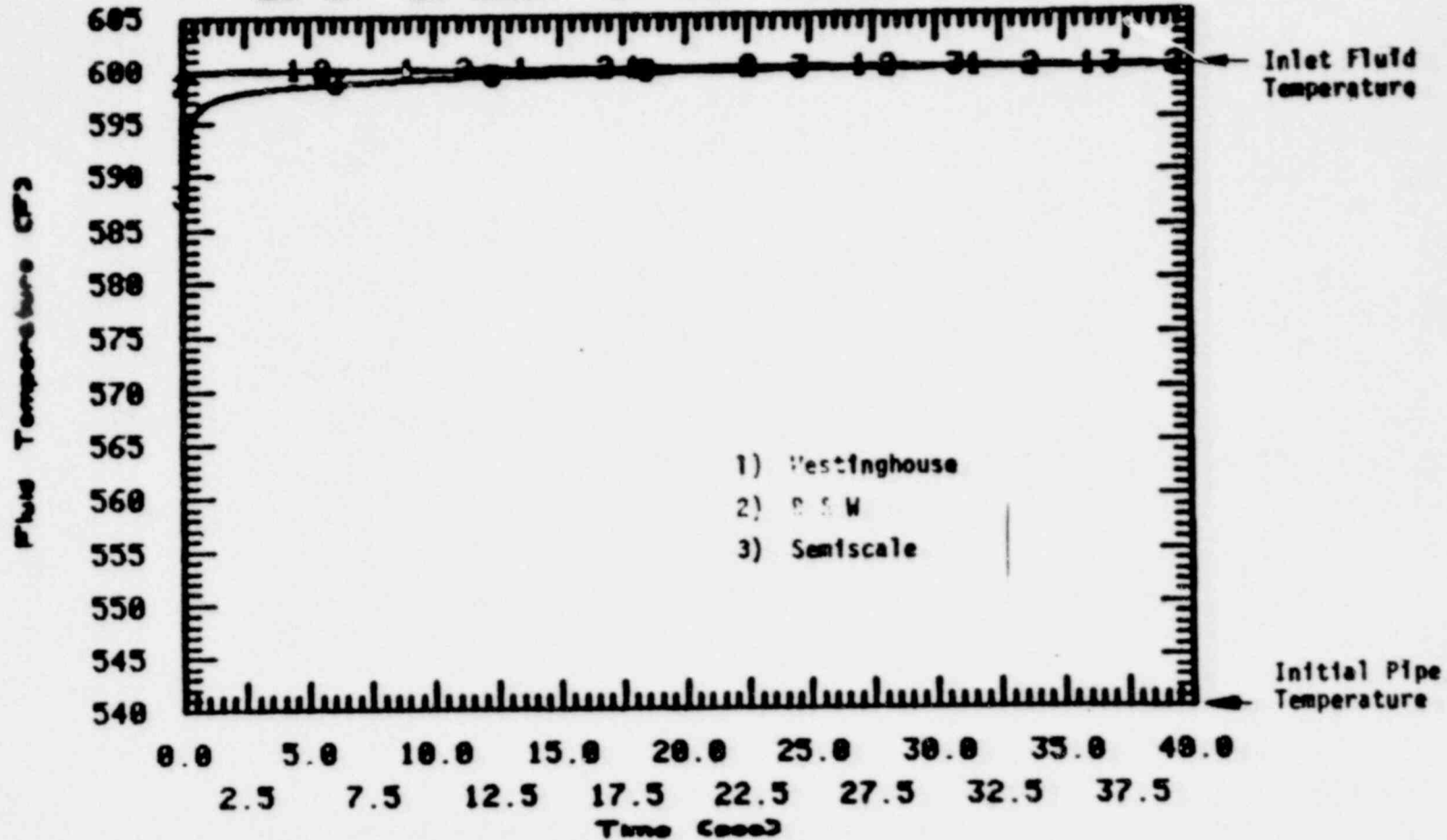


MODEL USED FOR PIPE WALL TRANSIT
 TIME CALCULATIONS

INFLUENCE OF PIPING ON FLUID TEMPERATURE

FOR A TEMPERATURE STEP AT THE INLET

FOR VEL. 80 OUTLET FLUID TEMPS.
1) W 2) B&W 3) SS DASH-D INLET



CORAL PIPE MODEL CALC.

241 4001
POOR ORIGINAL

1004 146

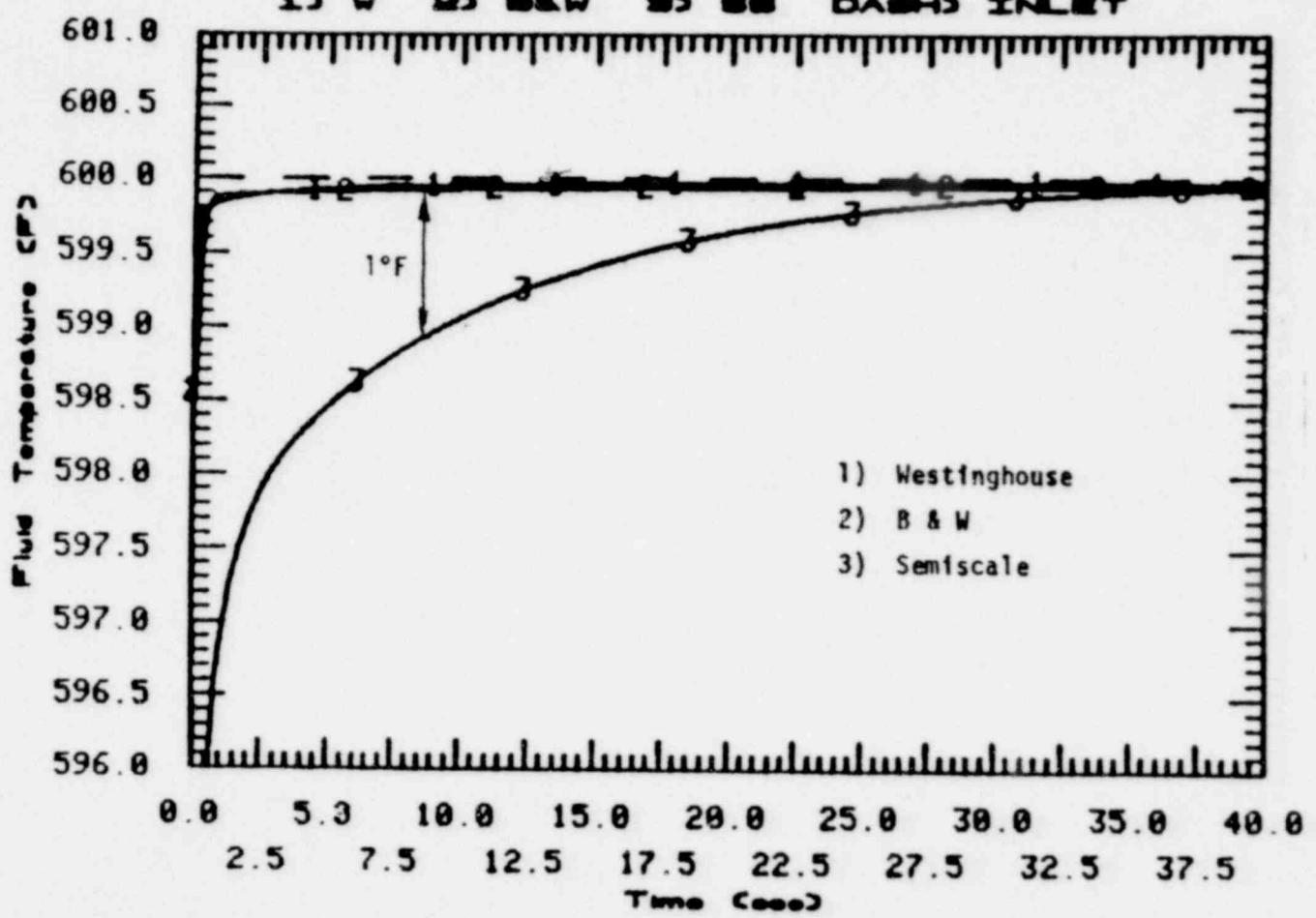
841 4001

POOR ORIGINAL

INFLUENCE OF PIPING ON FLUID TEMPERATURE

FOR A TEMPERATURE STEP AT THE INLET

PWR VS. SS OUTLET FLUID TEMPS.
1) W 2) B&W 3) SS DASH) INLET



CORAL PIPE MODEL CALCS.

1004 147

1004 148

FLOW REGIMES

INFLUENCE:

- WETTED AREA, PRESSURE DROP, CRITICAL FLOW

TREATMENT:

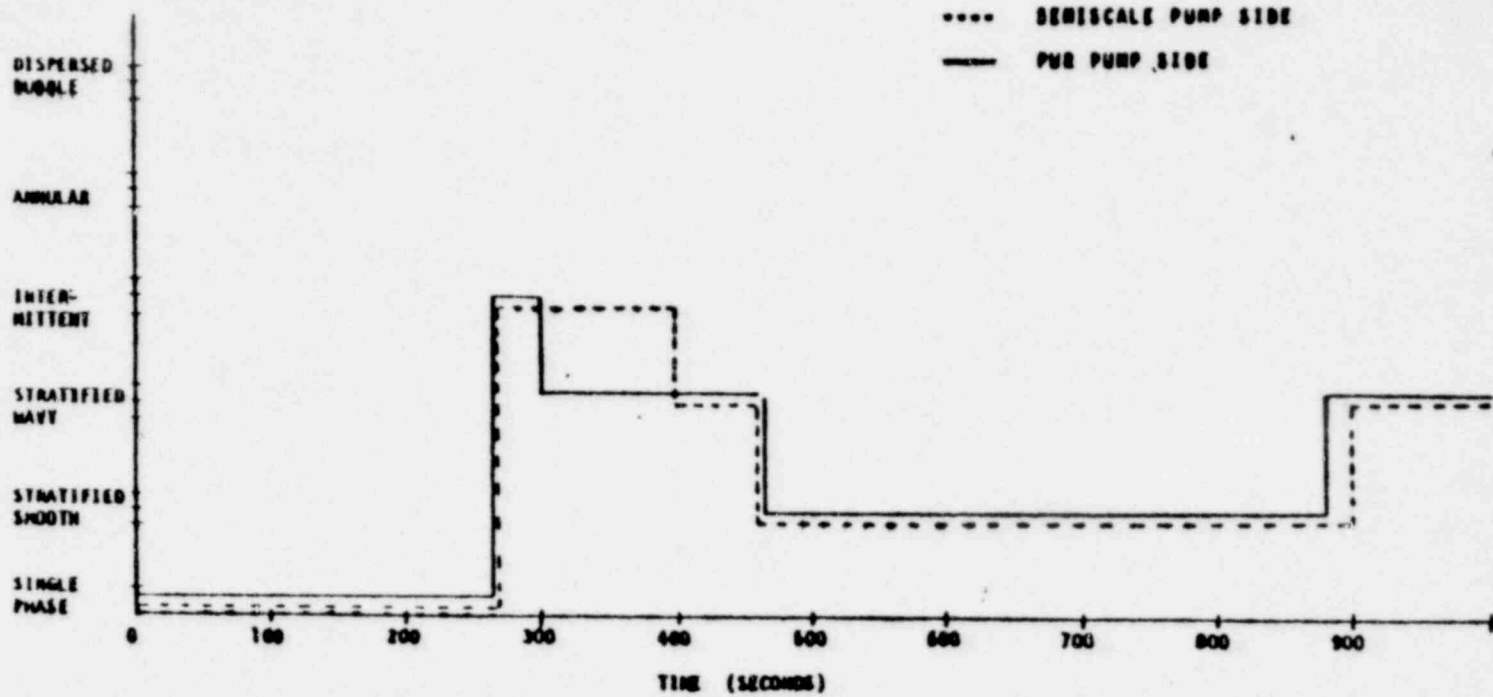
- FLOW REGIME CALCULATIONS
- DATA COMPARISON TO LARGER SCALE

1004 148

021 1001

POOR ORIGINAL

PREDICTED FLOW REGIMES IN SEMISCALE AND PWR COLD LEG PIPING DUKLER-TAITEL METHOD - PWR 4 IN. BREAK
AUDIT CALCULATION MASS FLUXES USED



1004 149

PAI 4001

LOOP PUMPS

INFLUENCE:

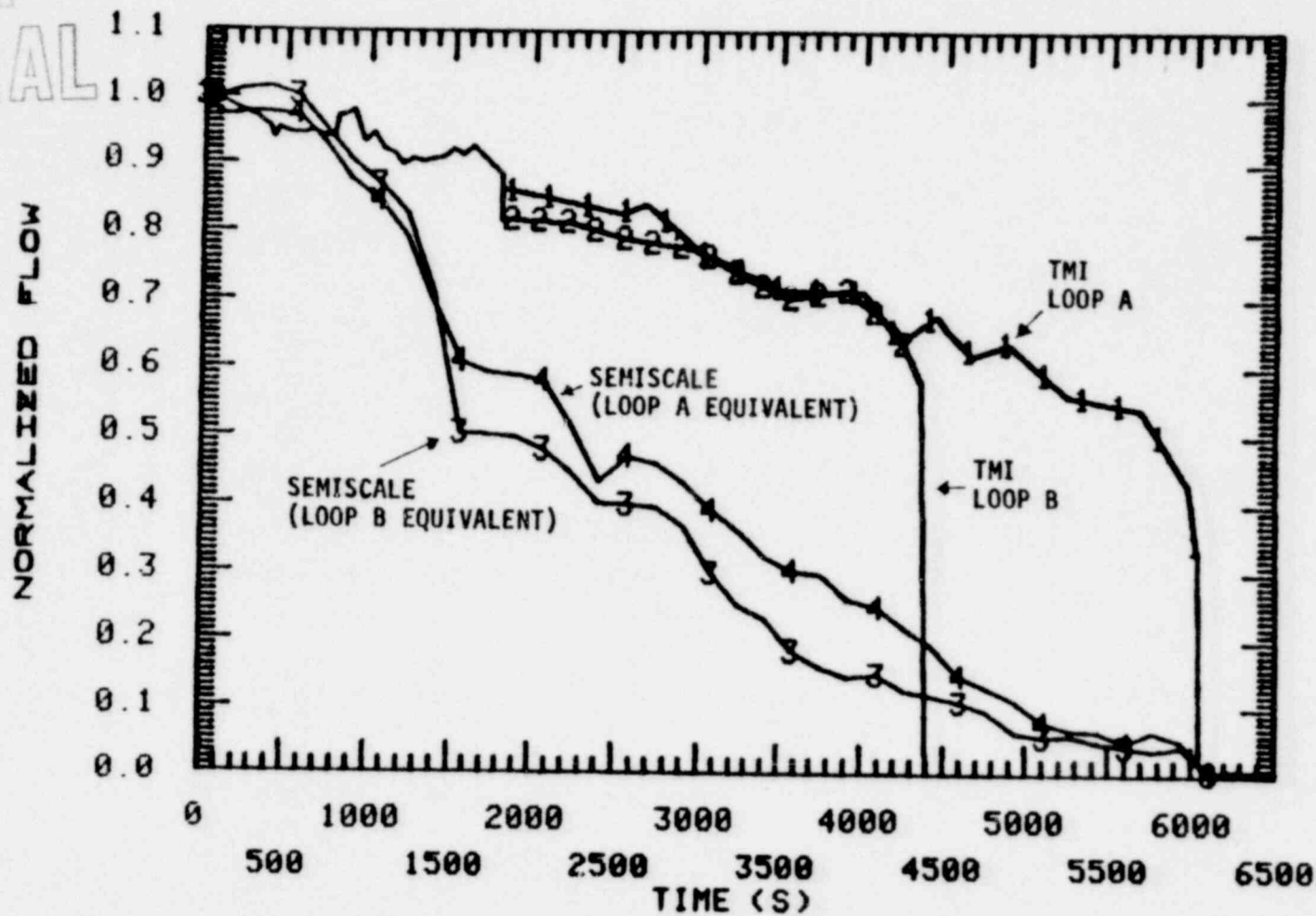
- LEAKAGE
- TWO-PHASE DEGRADATION

TREATMENT:

- PUMP REPLACEMENT, REDUCE LEAKAGE, PUMP LEAKAGE BACK INTO SYSTEM
- QUANTIFY PUMP PERFORMANCE

1004 150

COMPARISON OF NORMALIZED LOOP FLOWS FOR TMI AND THE SEMISCALE SIMULATION (TEST S-TMI-31)



POOR ORIGINAL

1004 151

1004 125

CRITICAL FLOW

INFLUENCE:

- SMALL SIZE OF SEMISCALE ORIFICE
- BOUNDARY LAYER EFFECTS

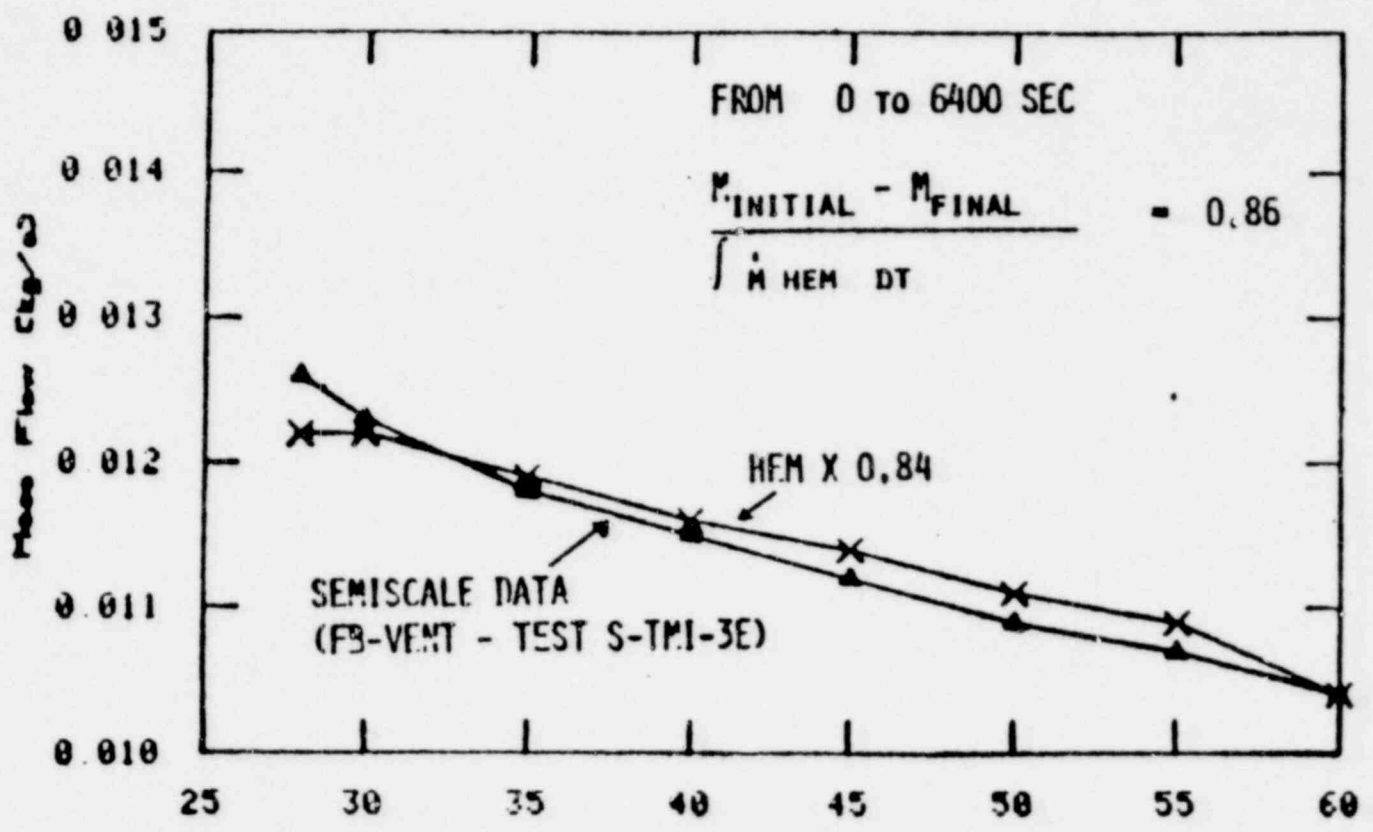
TREATMENT:

- RELATION OF MODELS TO RESULTS FROM PREVIOUS TESTS
- TESTS TO BE CONDUCTED IN LOFT TECHNICAL SUPPORT FACILITY

ACI 4001

POOR ORIGINAL

COMPARISON OF POV FLOW FROM SEMISCALE TEST S-TMI-3E AND FLOW PREDICTED FROM HOMOGENEOUS EQUILIBRIUM MODEL



1004 153

STEAM GENERATORS

INFLUENCE:

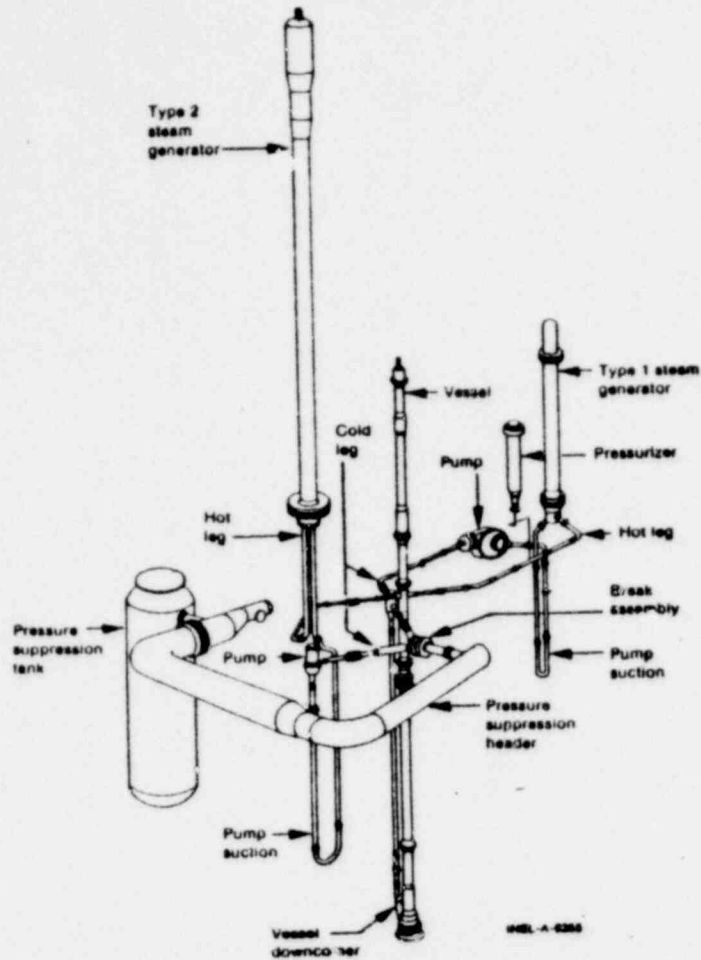
- OVERSIZED SECONDARY
- ELEVATION

TREATMENT:

- LONG TERM - NEW STEAM GENERATOR, REDUCE SECONDARY VOLUME
- NEAR TERM - REDUCE SECONDARY LEVEL
DRAIN SECONDARY
QUANTIFY ELEVATION INFLUENCE
DIMENSIONLESS NUMBER

POOR
ORIGINAL

SEMISCALE STEAM GENERATORS



SECONDARY WATER VOLUME
PRIMARY VOLUME

SURFACE AREA
PRIMARY VOLUME

HEIGHT (M)*

<u>INTACT LOOP</u>	<u>PWR THREE LOOPS</u>	<u>BROKEN LOOP</u>	<u>PWR ONE LOOP</u>
0.62	0.43	1.79	0.14
13.05	12.37	3.47	4.12
3.5	11.1	11.1	11.1

* COLD LEG CENTERLINE TO LOW TUBE SPILLOVER

DIMENSIONALITY

INFLUENCE:

- 1-D VERSUS POTENTIAL 2-3D INFLUENCE

TREATMENT:

- ANALYTICAL

821 1001

SUMMARY

- o VOLUME SCALING MOST APPROPRIATE
- o DISTORTIONS ARE PRESENT IN SYSTEM AS IN ALL SYSTEMS
- o DISTORTIONS ARE MANAGEABLE FROM CODE STANDPOINT
- o DATA WILL BE USEFUL FOR CODE BENCHMARKING / LOFT PHENOMENA IDENTIFICATION / QUANTIFICATION OF DISTORTIONS
- o SMALL BREAK DISTORTIONS NOT FULLY QUANTIFIED - FUTURE WORK REQUIRED

FUTURE SCALING WORK

- o JOINT SMALL BREAK SCALING ANALYSIS BETWEEN LOFT, SEMISCALE,
AND AUSTRIAN GOVERNMENT
- o IDENTIFICATION AND ANALYSIS OF SCALING CONCERNS FOR BOTH
ANTICIPATED TRANSIENT WITHOUT SCRAM AND LOSS-OF-FEEDWATER
EXPERIMENTS
- o DOCUMENTATION OF SCALING EFFORTS
- o ASSESSMENT OF THE EFFECTS OF DIMENSIONALITY

1004 158

1004 158

Rosztoczy

8/28/79

LOFT & SEMISCALE EXPERIMENTAL DATA NEEDS

TO BE PRESENTED AT
ACRS ECCS SUBCOMMITTEE MEETING

AUGUST 28, 1979

BY ZOLTAN R. ROSZTOCZY

1004 159

UPPER HEAD INJECTION (UHI)

- . NRR COMPLETED THE REVIEW OF UHI SYSTEMS IN 1977.
- . CONCLUSION OF REVIEW: THE AVAILABLE CALCULATIONAL TECHNIQUES ARE NOT SUFFICIENT TO QUANTIFY THE PROBABLE BENEFITS OR DRAWBACKS OF ADVANCED ECCS, LIKE UHI. WE DO NOT KNOW WHETHER UHI REPRESENTS AN IMPROVEMENT IN TERMS OF PUBLIC SAFETY.
- . BETTER, AND MORE COMPLETE EXPERIMENTAL INFORMATION IS ESSENTIAL:
 - (1) FOR THE UNDERSTANDING OF UHI PERFORMANCE
 - (2) FOR THE DEVELOPMENT AND VERIFICATION OF APPROPRIATE CALCULATIONAL TECHNIQUES.
- . NRR, THEREFORE, REQUESTED A TEST PROGRAM TO DEMONSTRATE AND EVALUATE UHI PERFORMANCE (MEMO FROM E.G. CASE TO S. LEVINE, DATED NOV. 8, 1976).
- . THIS IS STILL AN OUTSTANDING REQUEST. THE DEVELOPMENT OF ADVANCED ECCS IS A RATHER IMPORTANT ISSUE. THE TEST PROGRAM SHOULD START AS SOON AS POSSIBLE.

SMALL BREAK LOSS-OF-COOLANT ACCIDENT

- . COMPARISON OF CALCULATIONS AND SMALL BREAK INTEGRAL TEST DATA (SEMISCALE TEST S-02-6) INDICATED LARGE UNCERTAINTIES IN THE CALCULATIONS.
- . B&W LICENSING SUBMITTAL (SUMMER 1978) SHOWED THAT CALCULATIONS ARE RATHER SENSITIVE TO MODEST CHANGES IN THE MODEL.
- . SMALL BREAK CALCULATIONS ALSO INDICATED THAT THE CONSERVATISMS REQUIRED BY APPENDIX K HAVE ONLY A LIMITED EFFECT ON THE CALCULATIONS.
- . NRR REQUESTED A SET OF SMALL BREAK SEMISCALE TESTS TO
 - (1) EVALUATE THE UNCERTAINTIES OF SMALL LOCA CALCULATIONS.
 - (2) VALIDATE THE CALCULATIONAL METHODS.
- . NRC REQUIRED THAT REACTOR SUPPLIERS PERFORM PRETEST PREDICTION FOR THE FIRST SEMISCALE TEST.
- . FIRST TEST WAS COMPLETED IN DECEMBER 1978. FURTHER TESTS ARE OUTSTANDING.

SMALL BREAK LOSS-OF-COOLANT ACCIDENT

- . TMI-2 BROUGHT ATTENTION TO THE FACT THAT:
 - (1) VERY SMALL LOCA'S ARE OCCURRING WITH A HIGH FREQUENCY (4 EVENTS IN B&W PLANTS DURING 30 REACTOR YEARS OF OPERATION).
 - (2) PLANT RESPONSE TO VERY SMALL BREAKS DIFFERS FROM LARGER BREAK AND REQUIRES DIFFERENT OPERATOR ACTION (NATURAL CIRCULATION, OVERPRESSURIZATION, HIGH PRESSURIZER LEVEL).
 - (3) CALCULATIONAL TECHNIQUES MUST BE REVISED TO HANDLE VERY SMALL BREAKS.
 - (4) THERE IS INSUFFICIENT EXPERIMENTAL INFORMATION TO VALIDATE THE CALCULATIONS.
- . NRR REQUESTED DEMONSTRATION TESTS FOR
 - (1) THE ASSESSMENT OF PLANT BEHAVIOR FOLLOWING SMALL BREAKS.
 - (2) THE VALIDATION OF CALCULATIONAL METHODS.
 - (3) THE EVALUATION OF PLANT RECOVERY PROCEDURES.
- . NRC IS REQUIRING REACTOR SUPPLIERS TO PROVIDE PRETEST PREDICTIONS FOR SELECTED SMALL BREAK LOCA TESTS (NUREG-0578).
- . THE OUTSTANDING SMALL BREAK TEST REQUESTS ARE HIGH PRIORITY ITEMS AND SHOULD PROGRESS AS FAST AS PRACTICABLE.

SEMISCALE TEST S-07-6

- . TEST RESULTS INDICATED MULTIPLE VOIDING OF THE DOWNCOMER REPEATED WITH A CONSTANT FREQUENCY. RESULT: A 400 SEC. DELAY IN CORE QUENCHING.
- . CALCULATIONS PERFORMED WITH RELAP AND 1-D TRAC WERE UNABLE TO REPRODUCE THE OBSERVED PHENOMENON.
- . NRR REQUESTED EXPERIMENTAL AND ANALYTICAL STUDIES TO DECIDE WHETHER THE PHENOMENON IS TYPICAL FOR PWR'S AND WHETHER IT IS SIGNIFICANT FOR LICENSING.
- . PRESENT STATUS: (1) OBSERVED PHENOMENON IS REAL, IT CAN OCCUR IN LARGE SYSTEMS (PWR) AS WELL AS IN SMALL SYSTEMS (SEMISCALE); (2) LICENSING CODES AND RELAP ARE NOT APPROPRIATE TO HANDLE THIS PROBLEM; (3) PRELIMINARY 3-D TRAC CALCULATIONS SHOW DOWNCOMER VOIDING IN PWR'S DURING LOCA; AND (4) THE RELAP ANALYSIS INDICATES THE POSSIBILITY OF SUBCOOLED WATER BEING BYPASSED IN THE TESTS.
- . AN UPDATE TO THE BOARD NOTIFICATION WAS ISSUED ON MAY 30, 1979 INDICATING NRC CONCERN.
- . PWR VENDORS WERE REQUESTED EITHER TO SHOW THAT THIS PHENOMENON WILL NOT AFFECT THE SAFETY EVALUATION OF THEIR DESIGNS OR MODIFY THEIR EVALUATION MODEL TO ACCOUNT FOR THE PHENOMENON.
- . DEVELOPMENT OF AN ANALYTICAL CAPABILITY TO PREDICT THE DOWNCOMER MASS DEPLETION PHENOMENON IS NEEDED. CONFIRMATORY EXPERIMENTS ARE ALSO NEEDED. THIS IS A HIGH PRIORITY ITEM.

TRANSIENTS & ACCIDENTS

- . BWR TRANSIENT TESTS ON A NUCLEAR PLANT (PEACH BOTTOM) SHOWED LARGE ERRORS IN THE CALCULATIONAL TECHNIQUE USED FOR LICENSING.
- . NRR ISSUED A BOARD NOTIFICATION REQUEST EXPRESSING CONCERN.
- . CHALLENGING TEST RESULTS ON PWR TRANSIENTS ARE NOT AVAILABLE.
- . BOTH BWR & PWR TRANSIENT ANALYSIS METHODS NEED IMPROVEMENT AND VERIFICATION.
- . NRC IS CONSIDERING TO REQUIRE TRANSIENT TESTS ON TYPICAL BWR'S AND ON TYPICAL PWR'S.
- . EPRI HAS ASKED FOR TRANSIENT TESTS ON LOFT.
- . DEVELOPMENT OF A BALANCED TRANSIENT TEST PROGRAM UTILIZING NUCLEAR PLANTS AS WELL AS TEST APPARATUS (POSSIBLY LOFT) IS IMPORTANT.
- . SAFETY ANALYSIS METHODS USED FOR STEAM LINE BREAK AND FEEDLINE BREAK ANALYSIS ARE PRESENTLY UNDER NRC REVIEW.
- . ANALYSIS METHODS USED FOR STEAM GENERATOR TUBE RUPTURE HAVE NOT YET BEEN REVIEWED BY NRC.
- . EXPERIMENTAL VERIFICATION OF THESE METHODS IS RATHER LIMITED.
- . IMPORTANT ASPECTS OF ACCIDENT ANALYSES REQUIRING EXPERIMENTAL SUPPORT ARE:
 - STEAM GENERATOR DYNAMICS
 - MOISTURE CARRY-OUT
 - REALISTIC PLANT DYNAMICS

7/27/79

Broadus

BEACON DEVELOPMENT PROGRAM

C. R. BROADUS
ACRS MEETING
AUGUST 27, 1979



1004 165

TA 1 A001

BEACON

PURPOSE:

- TO PROVIDE A BEST-ESTIMATE CONTAINMENT ANALYSIS CAPABILITY TO PREDICT THE ACTUAL TRANSIENT.

1004 166

321 4001

POOR
ORIGINAL

LICENSING PHILOSOPHY

USE CONSERVATIVE CODES (HIGH P, T) WITH CONSERVATIVE
INPUT

- DO NOT KNOW THE DEGREE OF CONSERVATISM
- MAY COVER UP IMPORTANT PHENOMENA/PROBLEMS
- HIGH PRESSURES AND TEMPERATURES ARE NOT CONSERVATIVE
FOR SOME ANALYSES

1004 167

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1604 168

POOR
ORIGINAL

BEACON PHILOSOPHY

USE BEST ESTIMATE CODES TO PREDICT THE ACTUAL TRANSIENT

- COMPARISON WITH LICENSING ANALYSES TO DETERMINE THE DEGREE OF CONSERVATISM
- PREDICTION OF PHENOMENA WHICH MAY BE COVERED UP BY HOMOGENEOUS, EQUILIBRIUM CODES
- MAY BE USED FOR EUROPEAN LICENSING PHILOSOPHY (B.E. ± SAFETY FACTOR)
- MAY BE USED AS A GENERAL LICENSING TOOL TO GET A HANDLE ON NEW PROBLEMS.

1004 168

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BEACON METHODS

- BUILD FROM EXISTING CODES AND MODELS WHERE POSSIBLE:
 - LASL KFIX NUMERICAL SCHEME
(2-D, 2-PHASE, UNEQUAL VELOCITIES, UNEQUAL TEMPERATURES)
 - INEL HEAT 1 CONDUCTION SUBCODE
 - LASL/BNL WATER PROPERTIES
 - INEL DYNAMIC STORAGE ROUTINES
 - INEL INPUT AND PLOT PACKAGES

POOR
ORIGINAL

1004 169

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BEACON METHODS (CONTD . . .)

- DEVELOP NEW MODELS WHERE NECESSARY
 - AIR COMPONENT ADDED
 - MASS, MOMENTUM, ENERGY SOURCE
 - WALL FILM MODEL
 - HEAT TRANSFER CORRELATIONS
 - MESH COUPLING
 - VARIABLE MESH SPACING
 - PARTIAL FLOW BLOCKAGE
 - LUMPED PARAMETER MODEL
 - RESTART

- MODULAR IMPLEMENTATION

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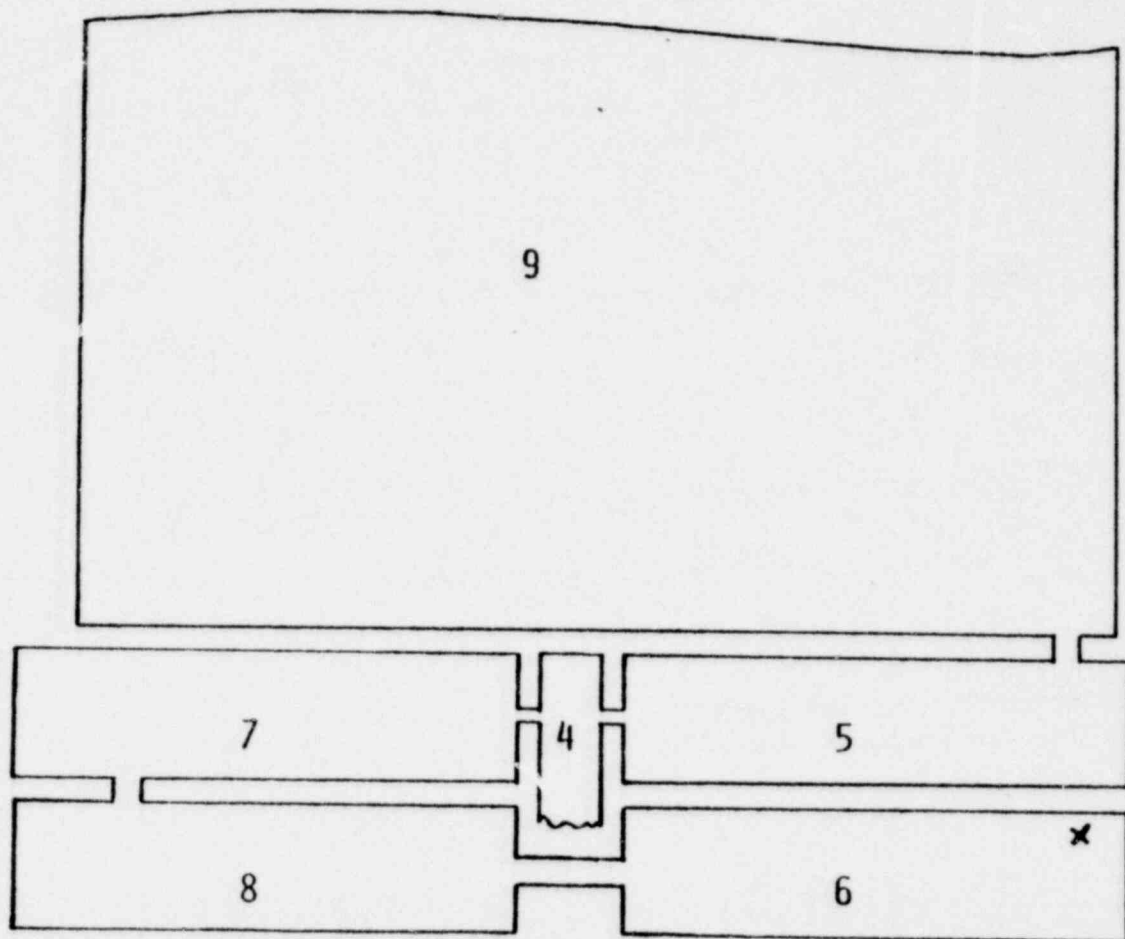
BEACON/MOD2A

- TWO - DIMENSIONAL
- NONEQUILIBRIUM
- TWO - COMPONENT (AIR AND WATER)
- TWO - PHASE (AIR/VAPOR AND LIQUID)
- COMPLEX GEOMETRIC MODELING
- SHORT TO INTERMEDIATE TERM
- HEAT TRANSFER
- WALL FILM

1004 171

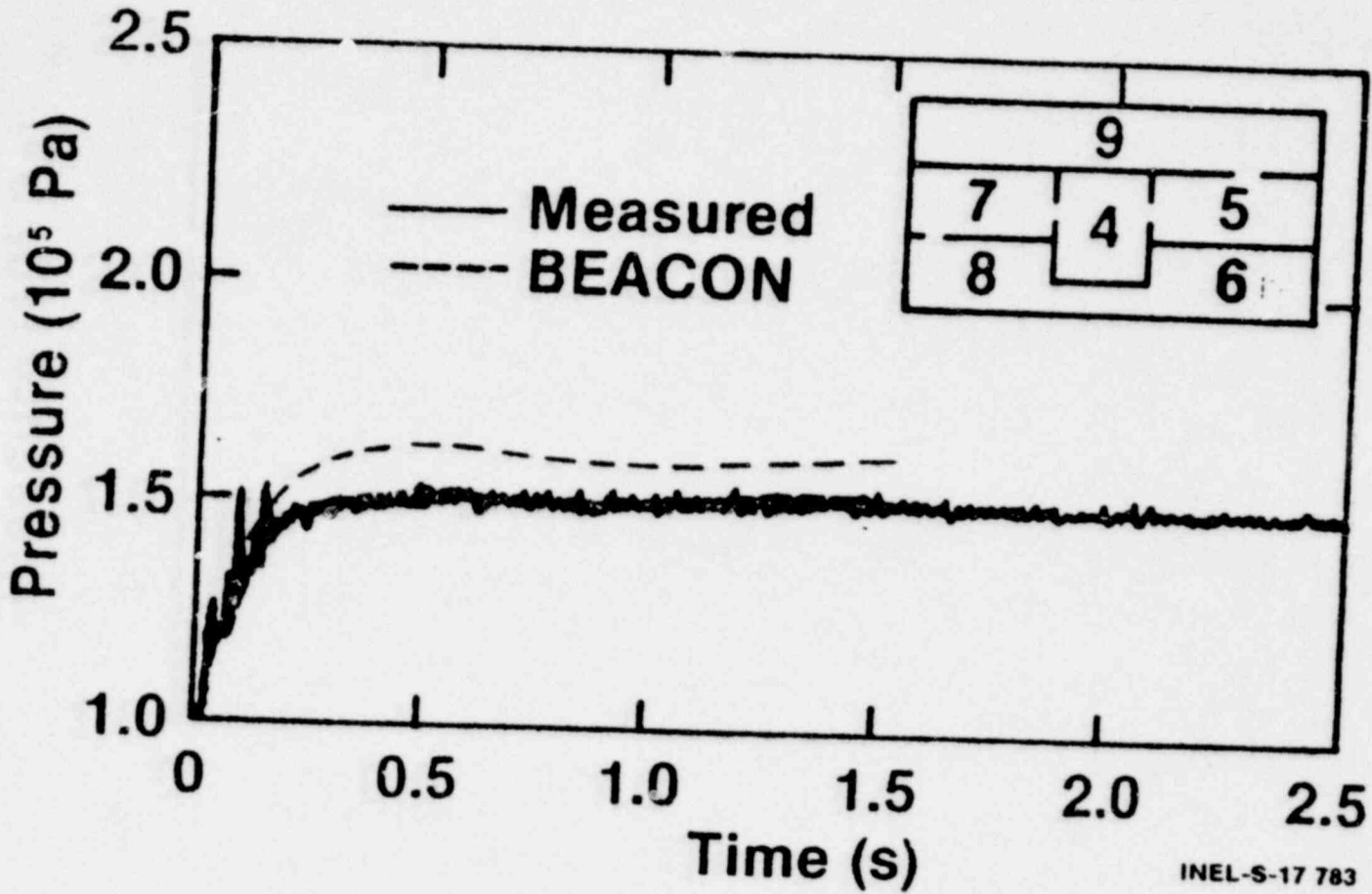
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BATTELLE - FRANKFURT D-15 TEST FACILITY



1004 172

Pressure in Room 6



INEL-S-17 783

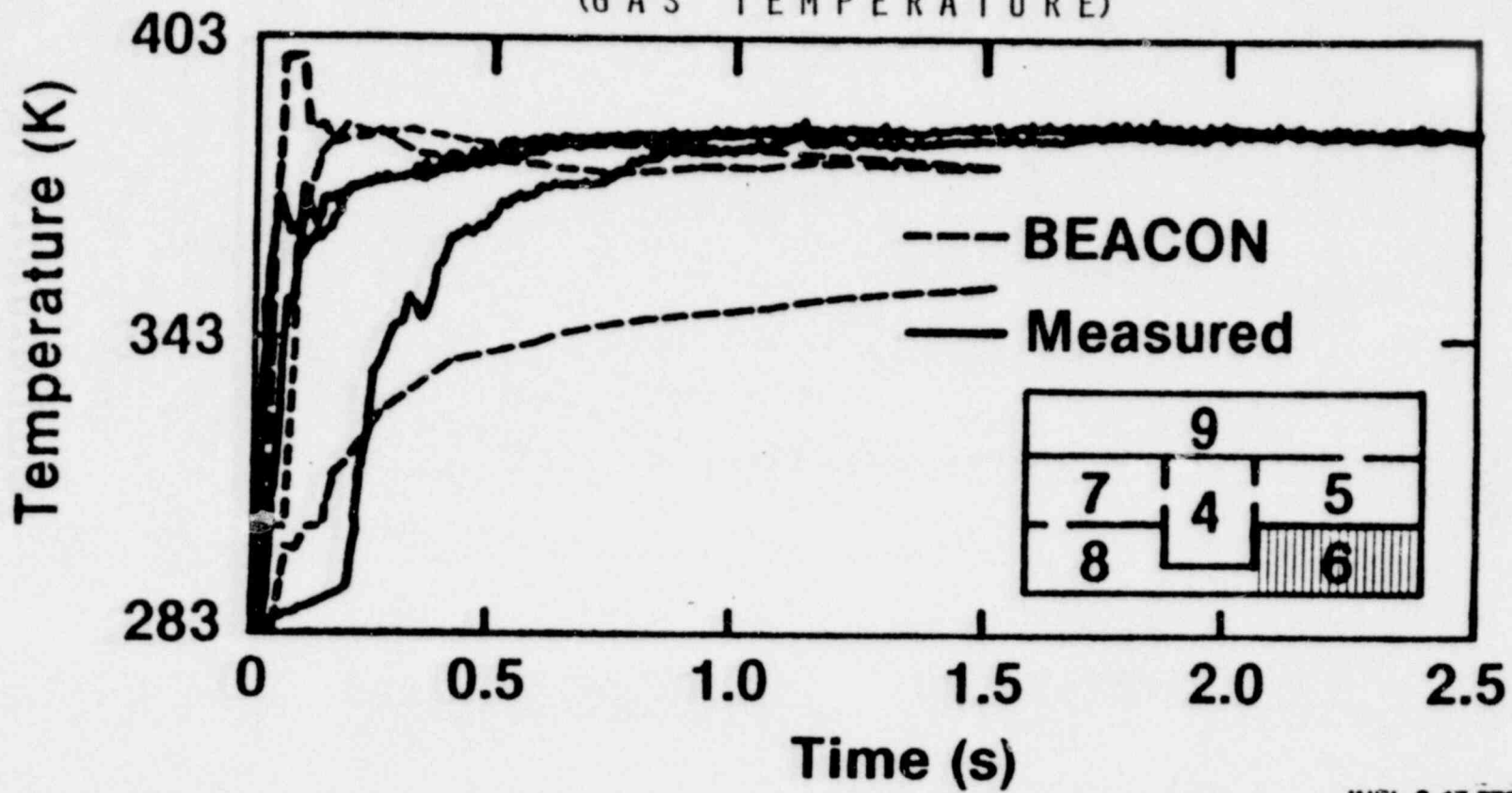
1004-173

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1004 174

Temperature in Room 6

(GAS TEMPERATURE)



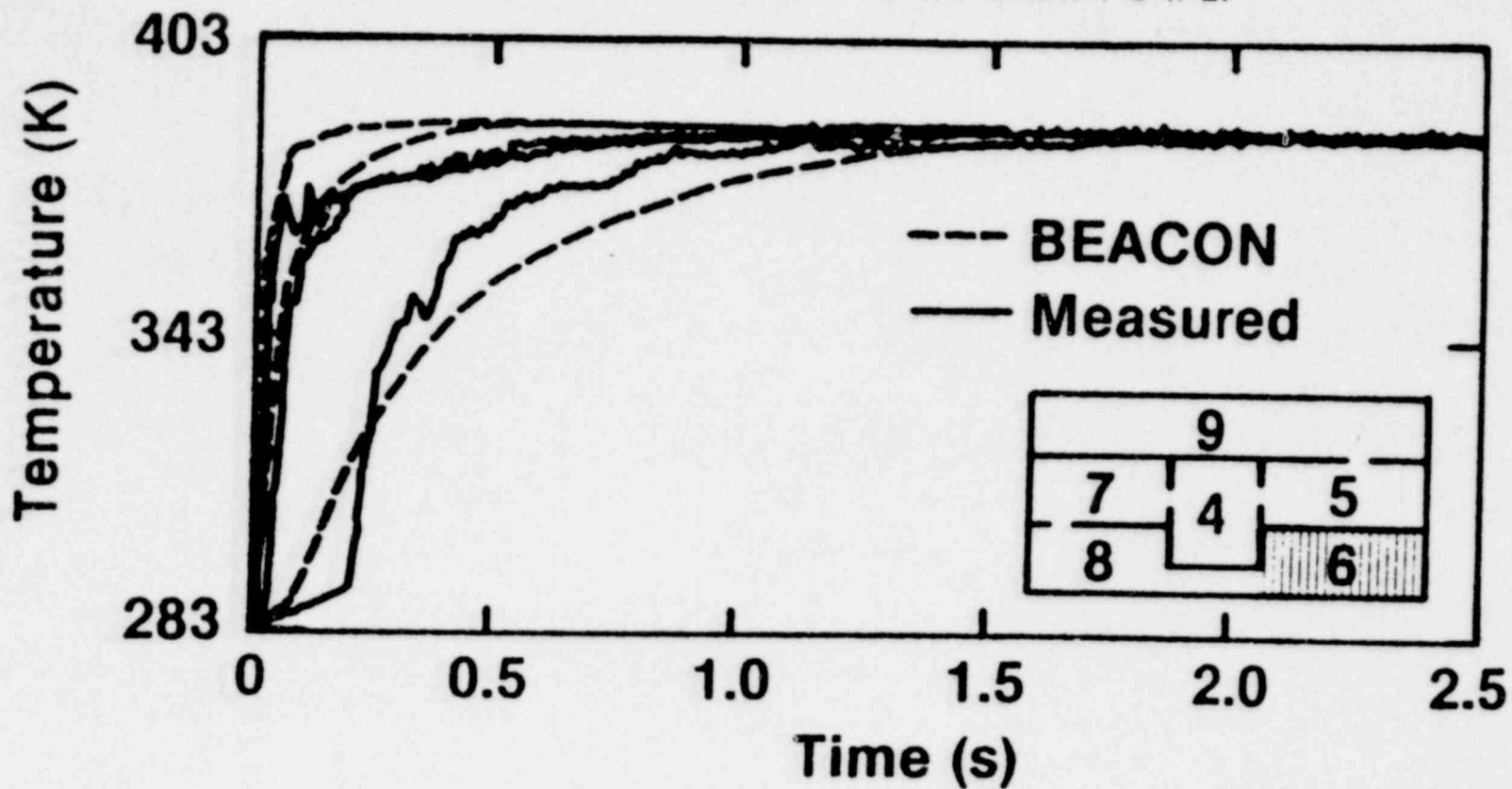
1004 174

INEL-6-17 778

1004 4001

Temperature in Room 6

(SATURATION TEMPERATURE)

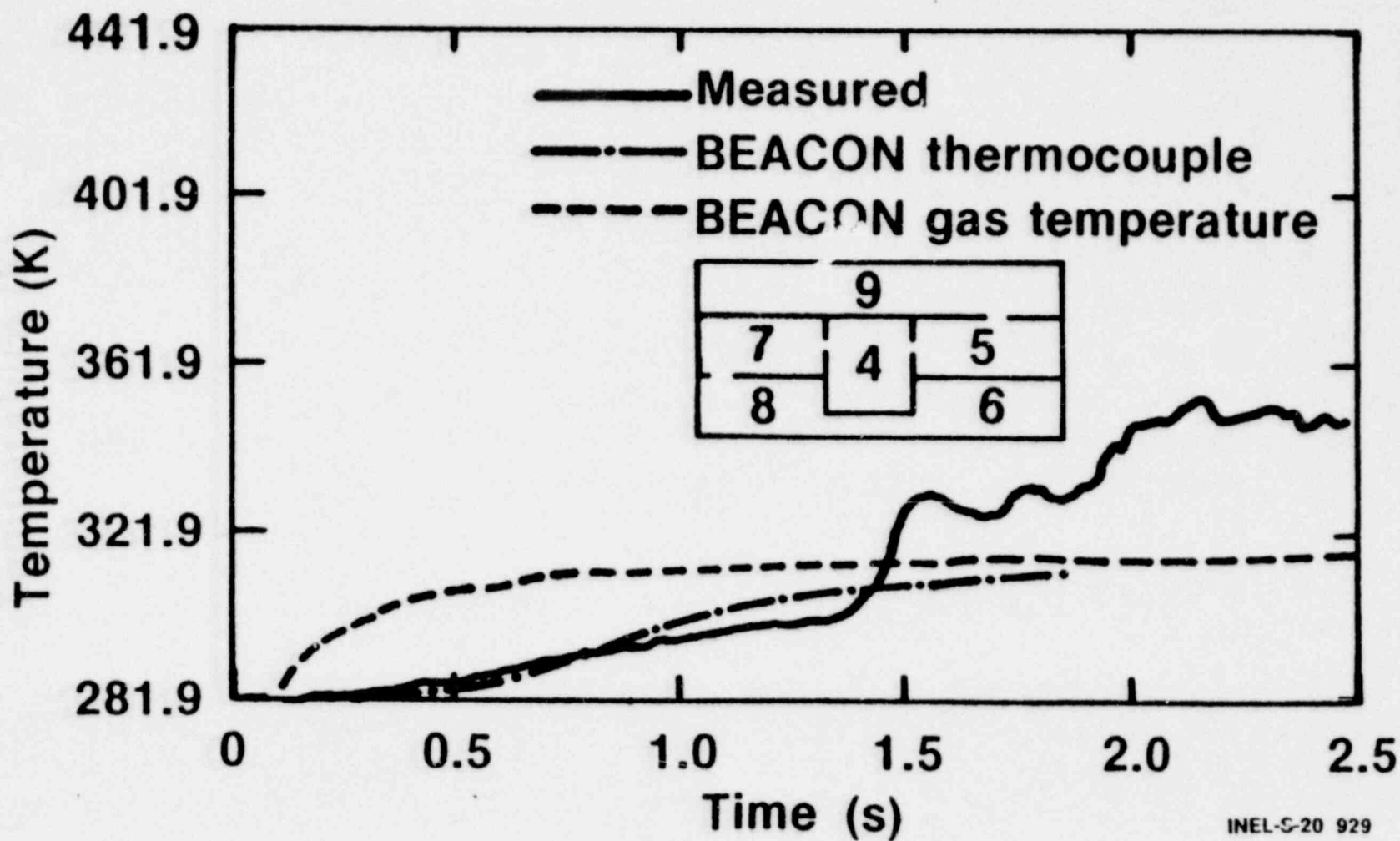


INEL-S-17 779

1004-175

49

Temperature in Room 7 Thermocouple Comparison



INEL-C-20 929

1004 176

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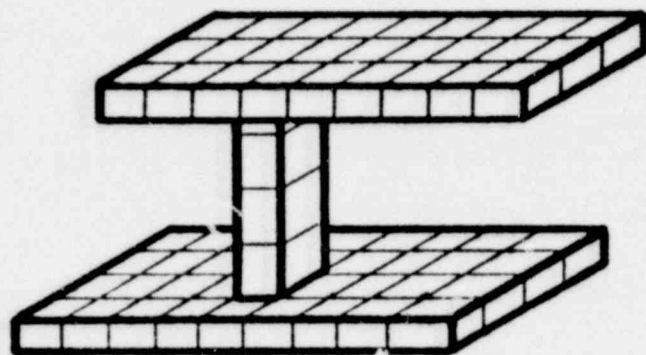
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CURRENT WORK

BEACON/MOD3 - NEW MODELS

- BEST ESTIMATE CORRELATIONS PACKAGE
- FORM AND FRICTION LOSS
- OUT - OF - PLANE COUPLING



1004 177

5

DEVELOPMENTAL ASSESSMENT PROGRAM

PURPOSE: DEFINE THE RANGE OF CONDITIONS/PROBLEMS TO WHICH BEACON IS APPLICABLE

METHODS: COMPARE TO DATA IN FOUR PHASES:

1. PROBLEM SETUP AND DOCUMENTATION
2. "BLIND" RUN
3. DATA COMPARISON
4. INPUT MODIFICATIONS WITHIN BEST-ESTIMATE INPUT LIMITS

PROBLEMS: SEPARATE EFFECTS PROBLEMS

- ENTRAINMENT/DEENTRAINMENT (DREXEL)
- 2-D SEPARATED FLOW (LAHEY)
- ETC.,

CONTAINMENT PROBLEMS

- BATTELLE - FRANKFURT C & D SERIES TESTS
- CVTR TEST
- ETC.,

8/28/79
Aahota

BEACON/MOD3
INTERPHASIC EXCHANGE RATES

M. S. SAHOTA
ACRS MEETING
AUGUST 27, 1979



1004 179

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BEACON/MOD3

INTERPHASIC EXCHANGE RATES

UNEQUAL VELOCITIES → INTERPHASIC DRAG

THERMODYNAMIC NONEQUILIBRIUM

→ HEAT AND MASS TRANSFER IN
THE PRESENCE OF INERT GAS

1004 180

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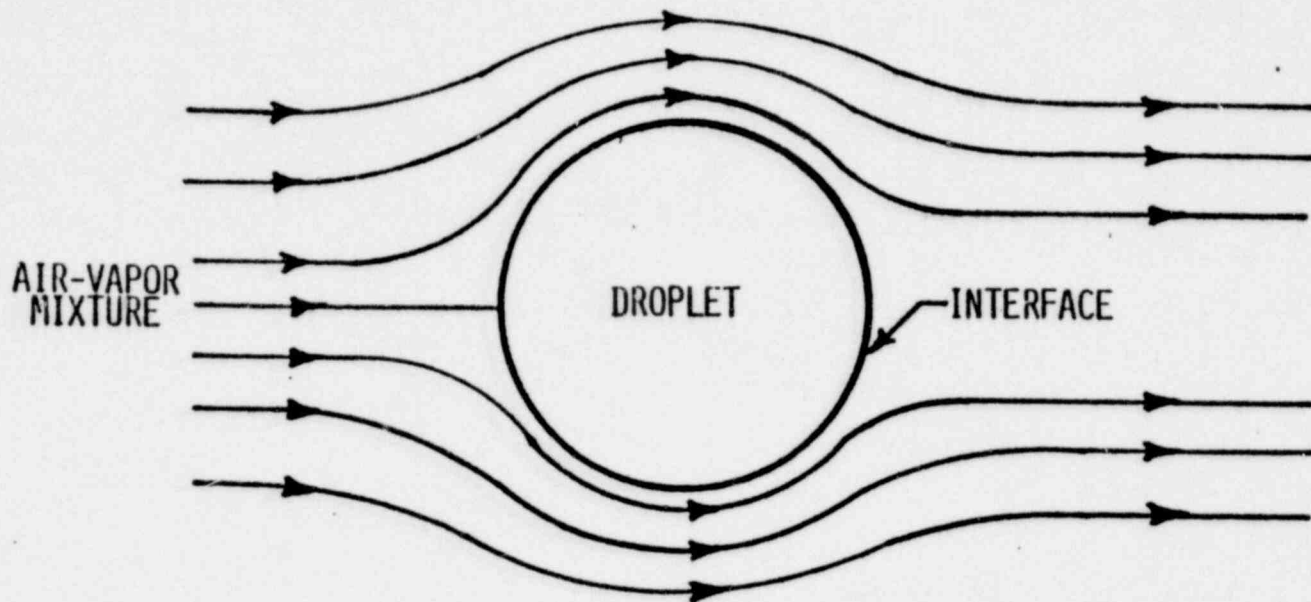
FLOW REGIMES

1. DISPERSED DROPLET FLOW ($\theta \rightarrow 1$)
2. BUBBLY FLOW ($\theta \rightarrow 0$)
3. VOID FRACTION CLOSE TO 0.5
4. OTHER FLOW REGIMES
 - (A) $0.5 \leq \theta \leq 1.0$
 - (B) $0 \leq \theta < 0.5$

1004 181

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1. DISPERSED DROPLET FLOW



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1004 182

(A) EXCHANGE RATES FOR A SINGLE DROPLET

$$\text{DRAG} = C_D \frac{1}{2} \rho_{\infty} U_{\infty}^2 \pi R^2$$

$$C_D = \begin{cases} \frac{24}{Re} & 1 + 0.15 Re^{0.687} & ; Re < 1300 \\ 0.4 & ; Re \geq 1300 \end{cases}$$

MASS BALANCE AT THE INTERFACE GIVES

$$\dot{M} = G_M \frac{w_{VI} - w_{V\infty}}{1 - v_{VI}}$$

ENERGY BALANCE AT THE INTERFACE YIELDS

$$\dot{E} = G_H (H_{MI} - H_{M\infty}) + \dot{M} H_{MI}$$

(LE = 1)

SPECIES EQUATION SOLUTION GIVES

$$\frac{G_M}{G_H} = \frac{G_H}{G_H} = \frac{1 - w_{VI}}{w_{VI} - w_{V\infty}} \quad \text{LN} \left(\frac{1 - w_{V\infty}}{1 - w_{VI}} \right)$$

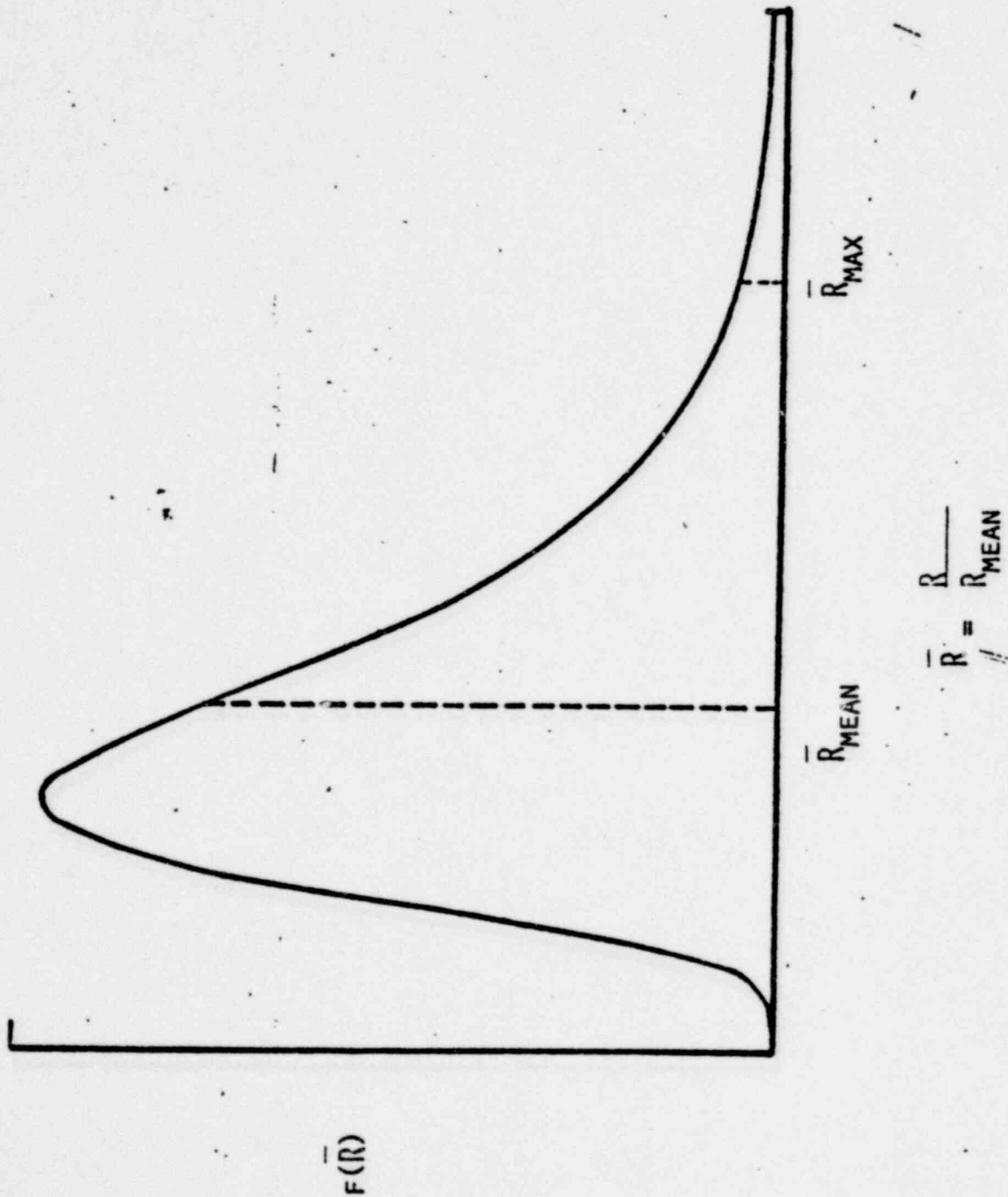
FLOW PAST A SPHERE

$$Nu = 2.0 + 0.60 Pr^{1/3} Re^{1/2}$$

881 A001

1004 184

(B) DROPLET SIZE DISTRIBUTION



(C) MACROSCOPIC EXCHANGE RATES

$$\text{TOT/L DRAG} = \int_0^{\infty} C_D \frac{1}{2} \rho_{\infty} U_{\infty}^2 \pi R^2 \cdot N_F(\bar{R}) d\bar{R}$$

$$\text{NU}_{\text{EFF}} = \frac{1}{A} \int_0^{\infty} \text{Nu} \cdot 4\pi R^2 N_F(\bar{R}) d\bar{R}$$

281 4001

2. BUBBLY FLOW

(A) EXCHANGE RATES FOR A SINGLE BUBBLE

DRAG - SAME APPROACH AS FOR DROPLET

$$\dot{E} = G_L (H_L - H_{L1}) + \dot{M} H_{L1} = \dot{M} H_{V1}$$

$$\dot{M} = \frac{G_L (H_L - H_{L1})}{H_{FG}}$$

(B) BUBBLE SIZE DISTRIBUTION

(C) MACROSCOPIC EXCHANGE RATES



SAME APPROACH AS FOR DROPLETS

1004 187

3. VOID FRACTION CLOSE TO 0.5

LIQUID ↓ ↑ AIR-VAPOR MIXTURE

(A) EXCHANGE FLUXES - DIFFUSION LIMITED IN GAS PHASE

$$\tau_1 = 1/2 F \rho_w U_w^2$$

$$ST = \frac{(F/2)^{1/2}}{[5 PR + 5 LN (5 PR + 1) + (F/2)^{1/2} - 14]}$$

(B) MACROSCOPIC EXCHANGE RATES

- MULTIPLY BY AREA

188 1004

1004 188

4. OTHER FLOW REGIMES

(A) $0.5 \leq \theta \leq 1.0$

VOID FRACTION WEIGHTED DRAG, HEAT AND MASS TRANSFER
COEFFICIENTS

(B) $0 \leq \theta < 0.5$

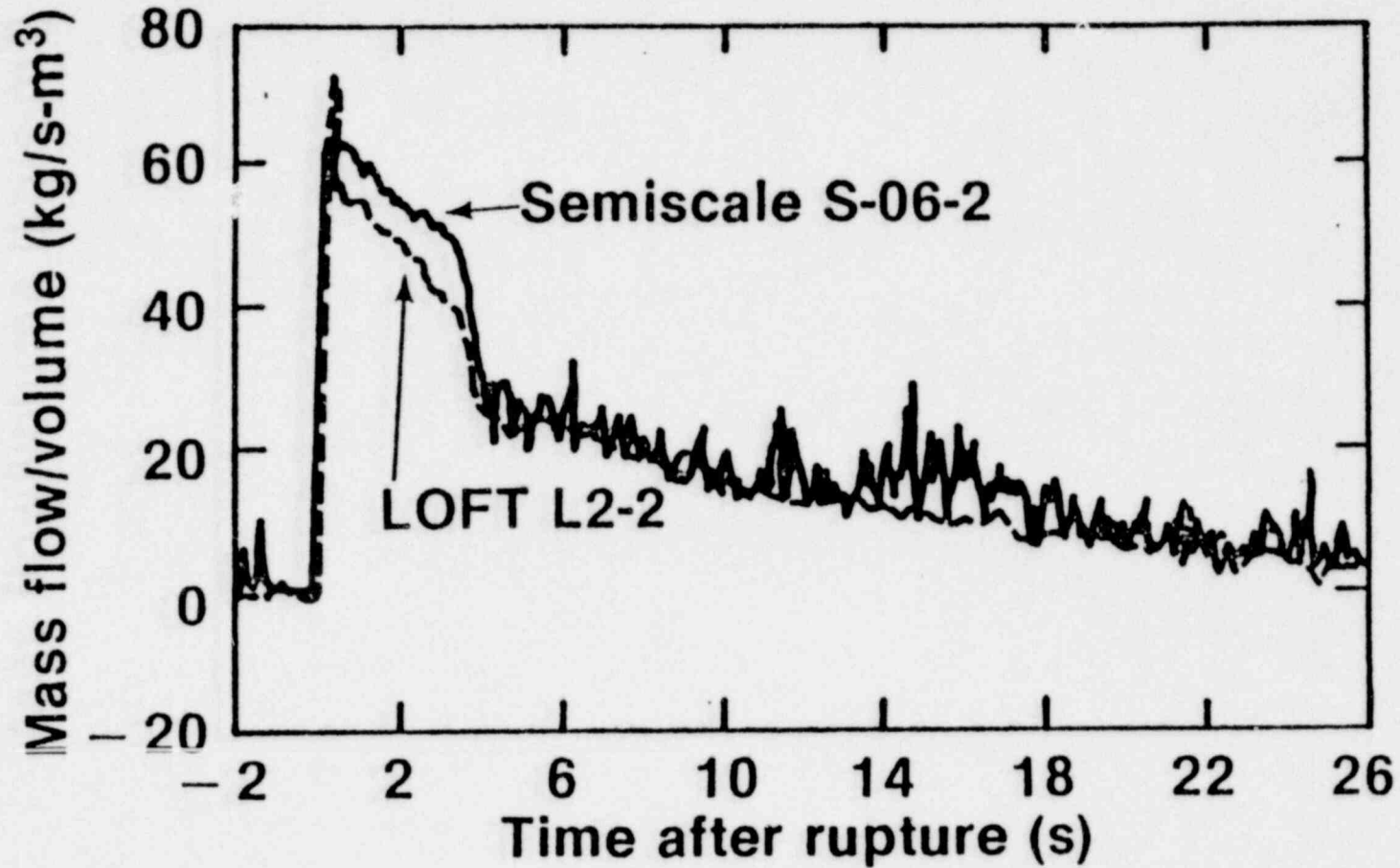
FORCE AND ENERGY BALANCES AT THE INTERFACE

$$\frac{1}{K} = \frac{1}{K_G} + \frac{1}{K_L}$$

Supplement Larson
8/20/79

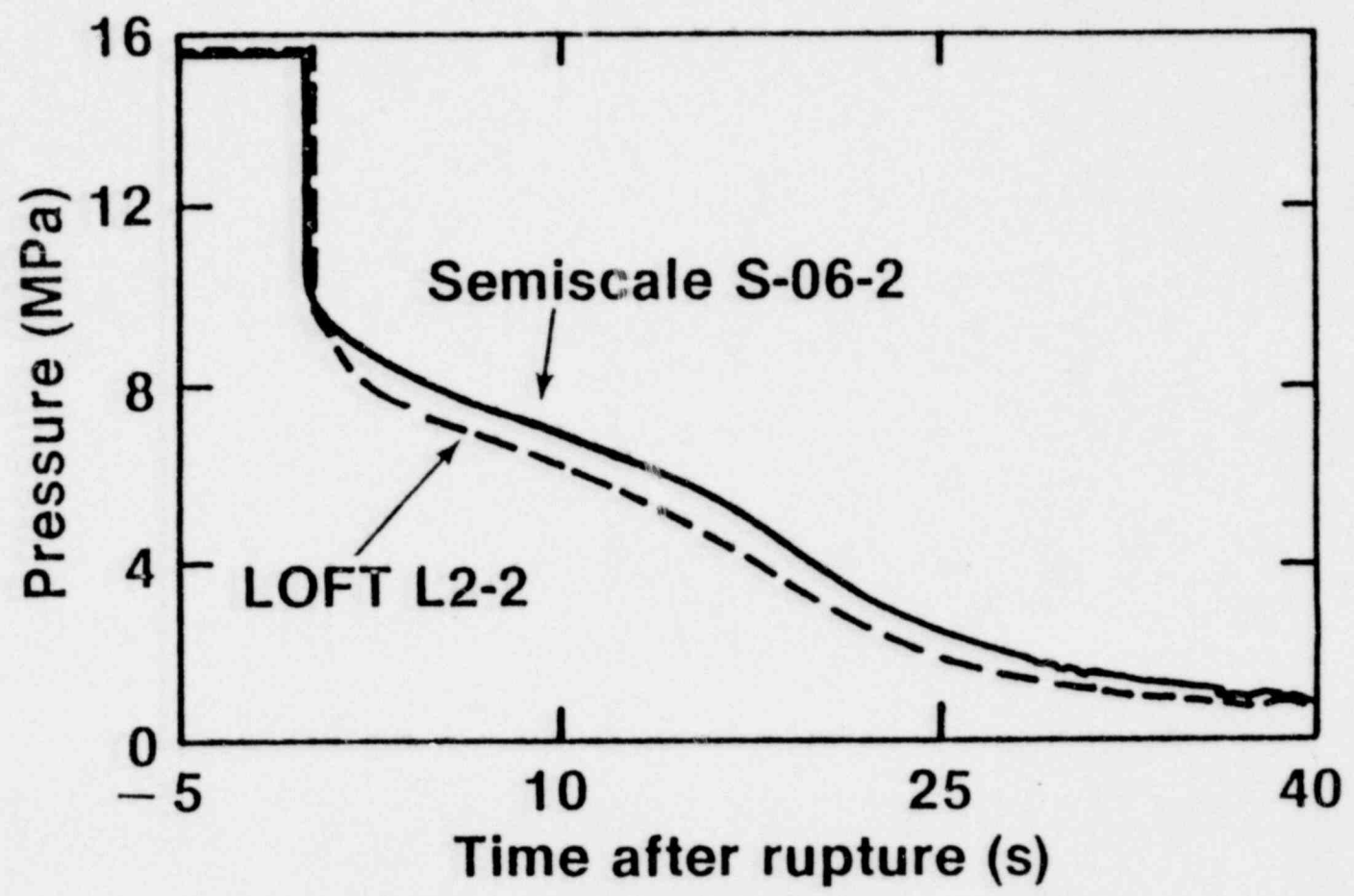
1004 190

Broken Cold Leg Mass Flow (Normalized to System Volume)



INEL-S-20 232

System Pressure



PRESSURIZER SURGE LINE

INFLUENCE:

- SIZE
- HYDRAULIC RESISTANCE

TREATMENT:

- SCALED RESISTANCE

COMPARISON OF PWR AND SEMISCALE SURGE LINES

	PWR [A]	SEMISCALE [B]
DIAMETER (IN)	8.5	0.37
AREA (FT ²)	3.941×10^{-1}	7.467×10^{-4}
R' (LBF-S ² / LBM-IN ² -FT ³)	8.617×10^3 [C]	6.200×10^3 [D]
D ⁺	1.333×10^2	5.804
J _G (FT/S) [E]	4.206×10^1	2.15×10^1
J _G CRITICAL (FT/S)	1.135×10^1	2.369

[A] B&W

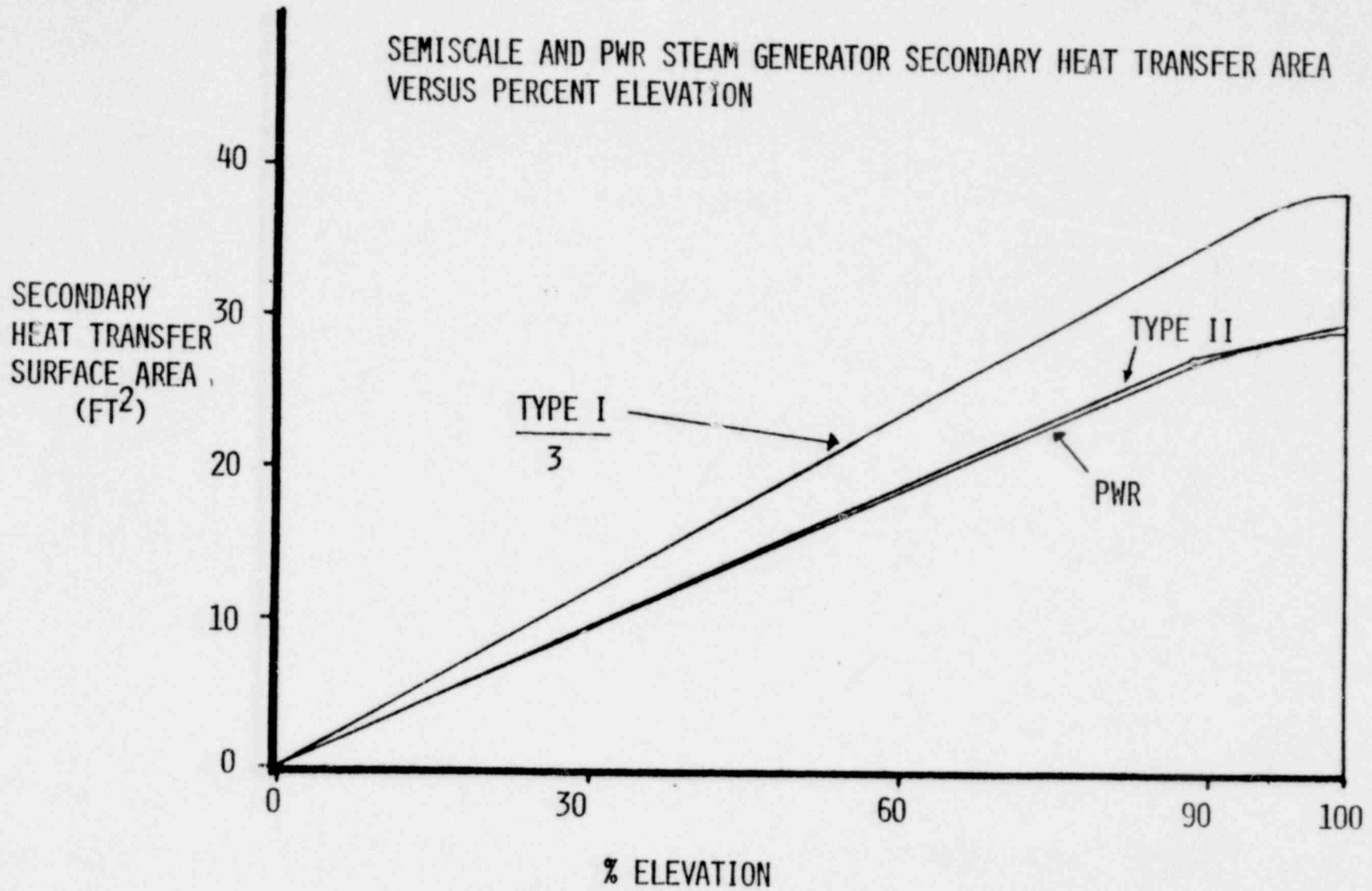
[B] VALUES LISTED ARE FOR THREE MILE ISLAND SIMULATIONS

[C] SCALED TO SEMISCALE

[D] CALCULATED

[E] FOR CASE OF POV LIQUID DISCHARGE AT PRESSURE OF 1000 PSIA

SEMISCALE AND PWR STEAM GENERATOR SECONDARY HEAT TRANSFER AREA
VERSUS PERCENT ELEVATION



Johnsen
8/28/79

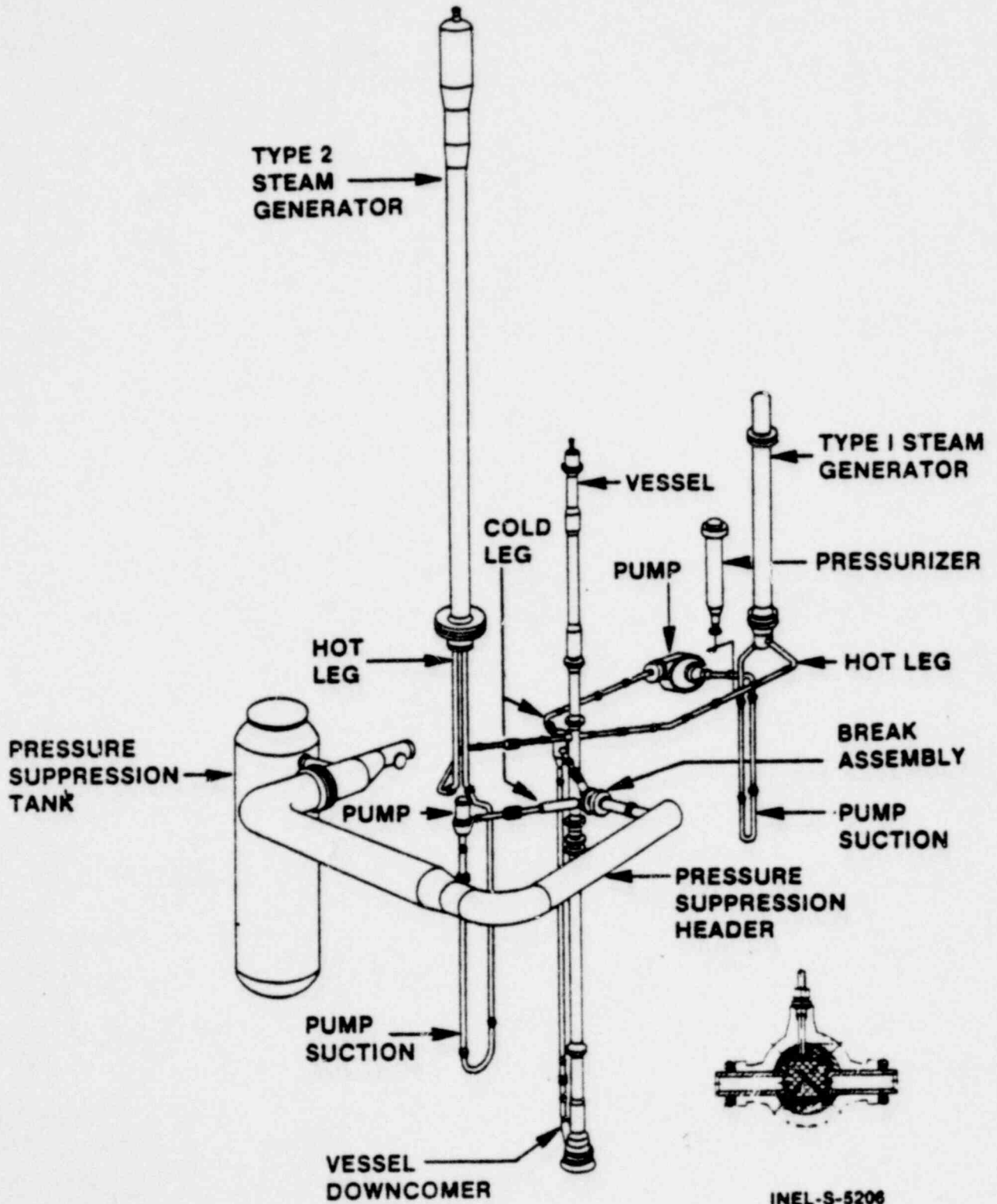
SEMISCALE PROGRAM ACCOMPLISHMENTS

G. W. JOHNSEN
ACRS MEETING
AUGUST 27, 1979



1004/195

MOD-3 SEMISCALE COLD LEG BREAK ASSEMBLY



EXPERIMENTS CONDUCTED

- . SERIES 7 - MOD-3 BASELINE
- . SMALL BREAK - STANDARD PROBLEM
- . LOWER PLENUM INJECTION
- . TMI SIMULATIONS
- . PLANNING TESTS

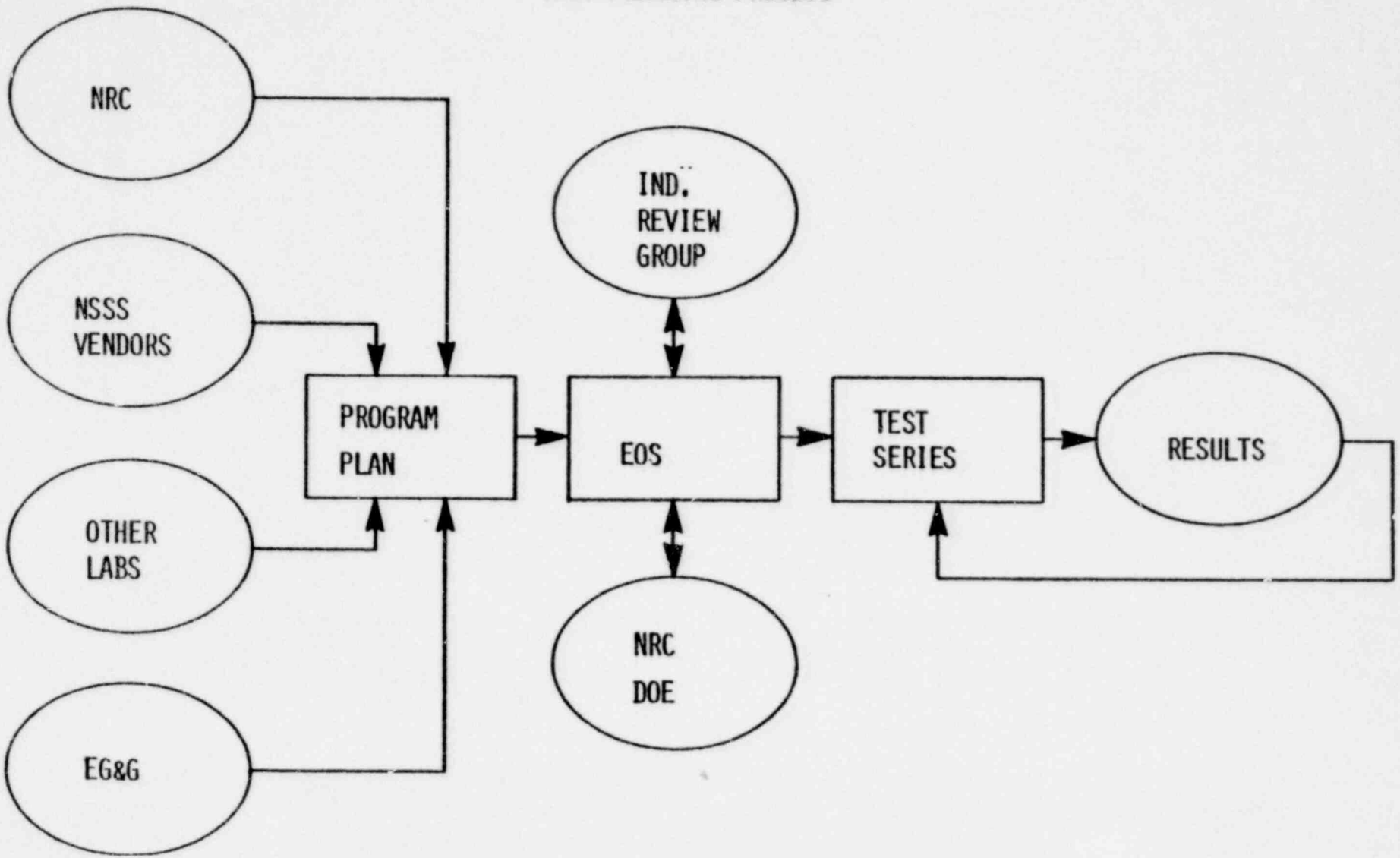
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1004 197 /4

PP1 4001

1004 198 72

TEST PLANNING PROCESS



891 4001

SEMISCALE'S ROLE IN SAFETY RESEARCH

- . CODE ASSESSMENT
- . BASIC SYSTEM PHENOMENA
- . EFFECTS OF SCALE

1004 199

13

TEST SERIES 7 RESULTS

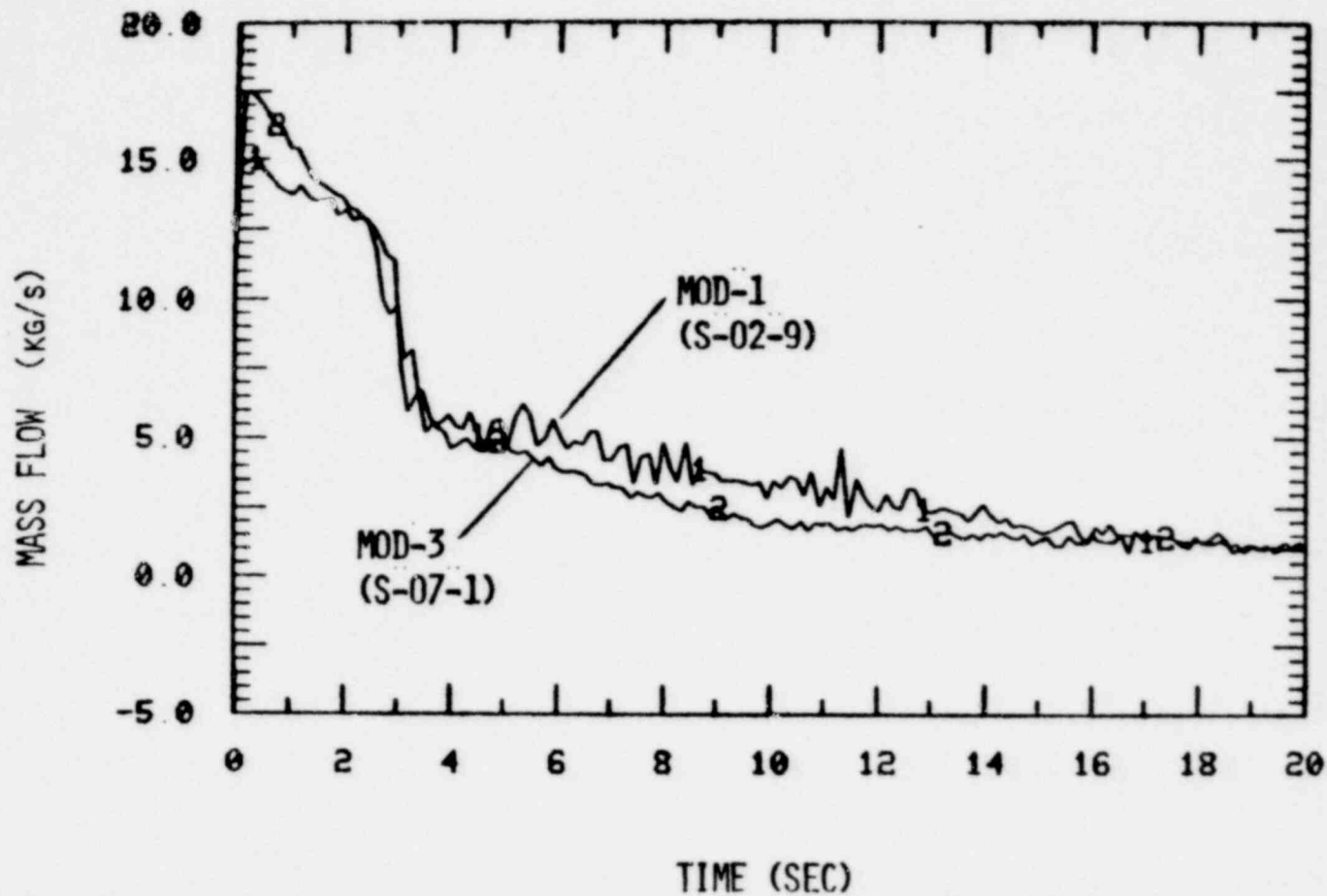
- . SCALING (MOD-1 VS. MOD-3)
- . MASS DEPLETION

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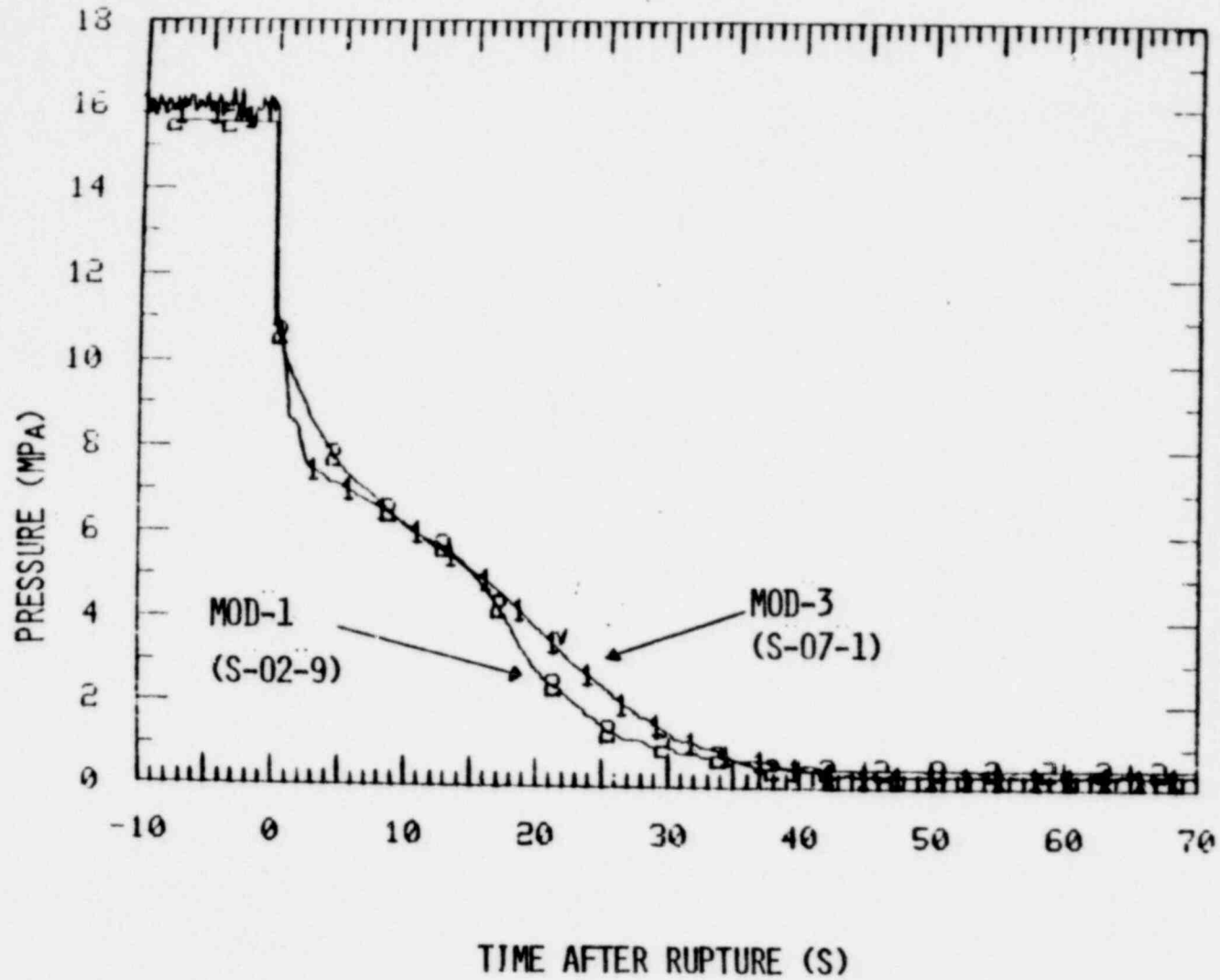
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COMPARISON OF COLD LEG BREAK FLOW

MOD-1 VERSUS MOD-3



COMPARISON OF DEPRESSURIZATION
MOD-1 VERSUS MOD-3



EOS 4001

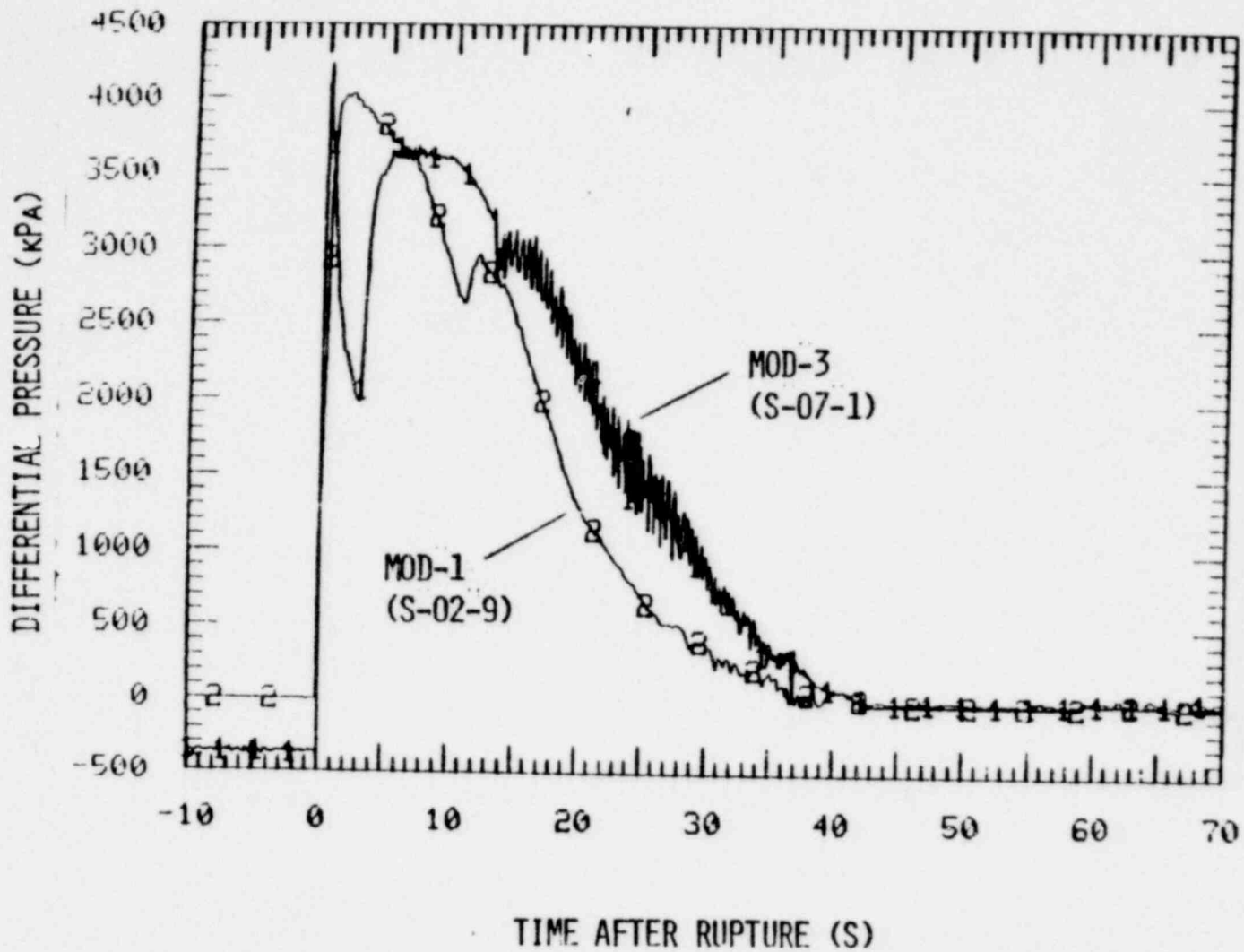
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POOR ORIGINAL

COMPARISON OF PRESSURE DIFFERENTIAL

MOD-1 BROKEN LOOP PUMP SIMULATOR VERSUS MOD-3 ACTIVE PUMP



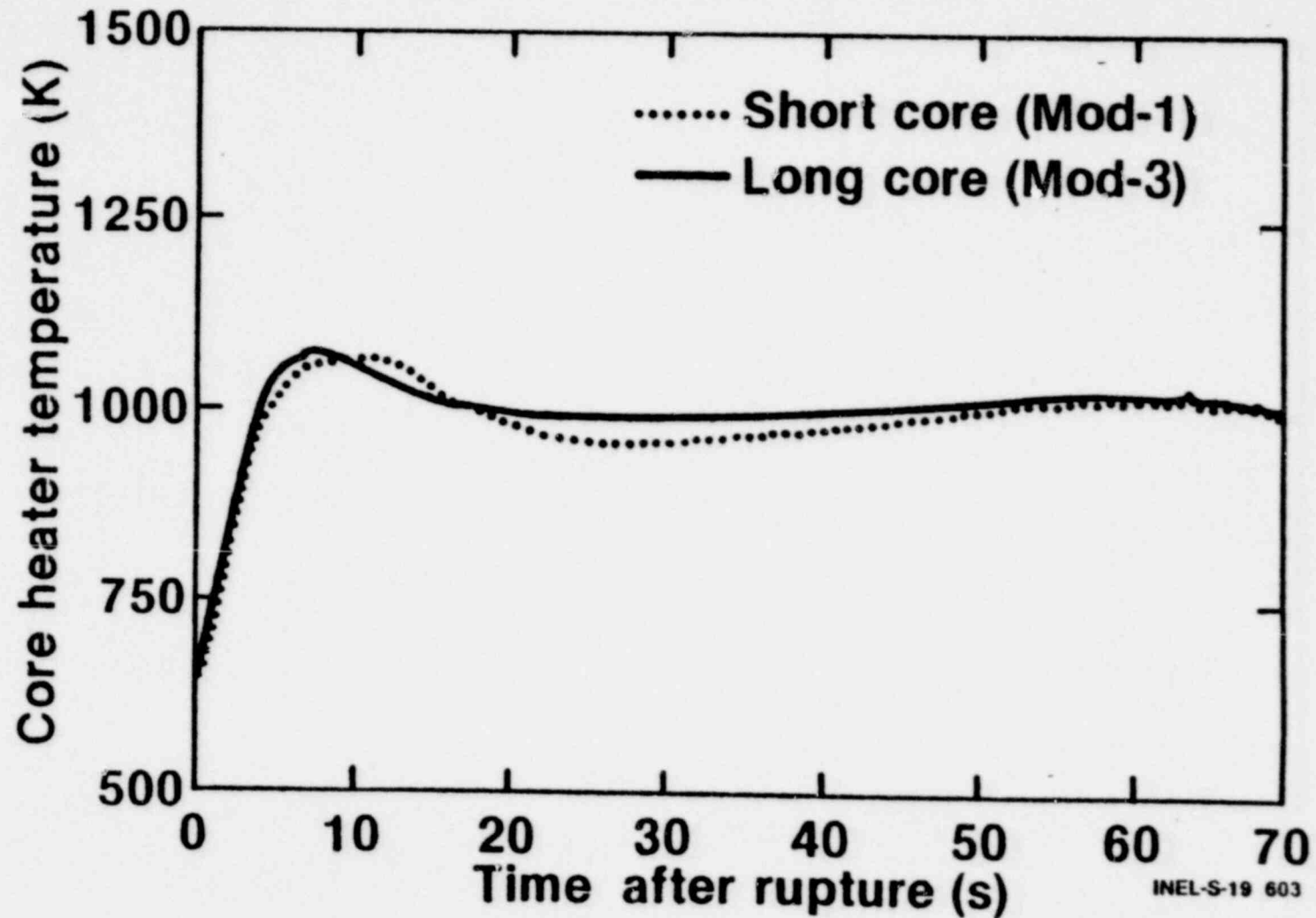
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Measured Rod Temperatures at Midplane of Semiscale Core

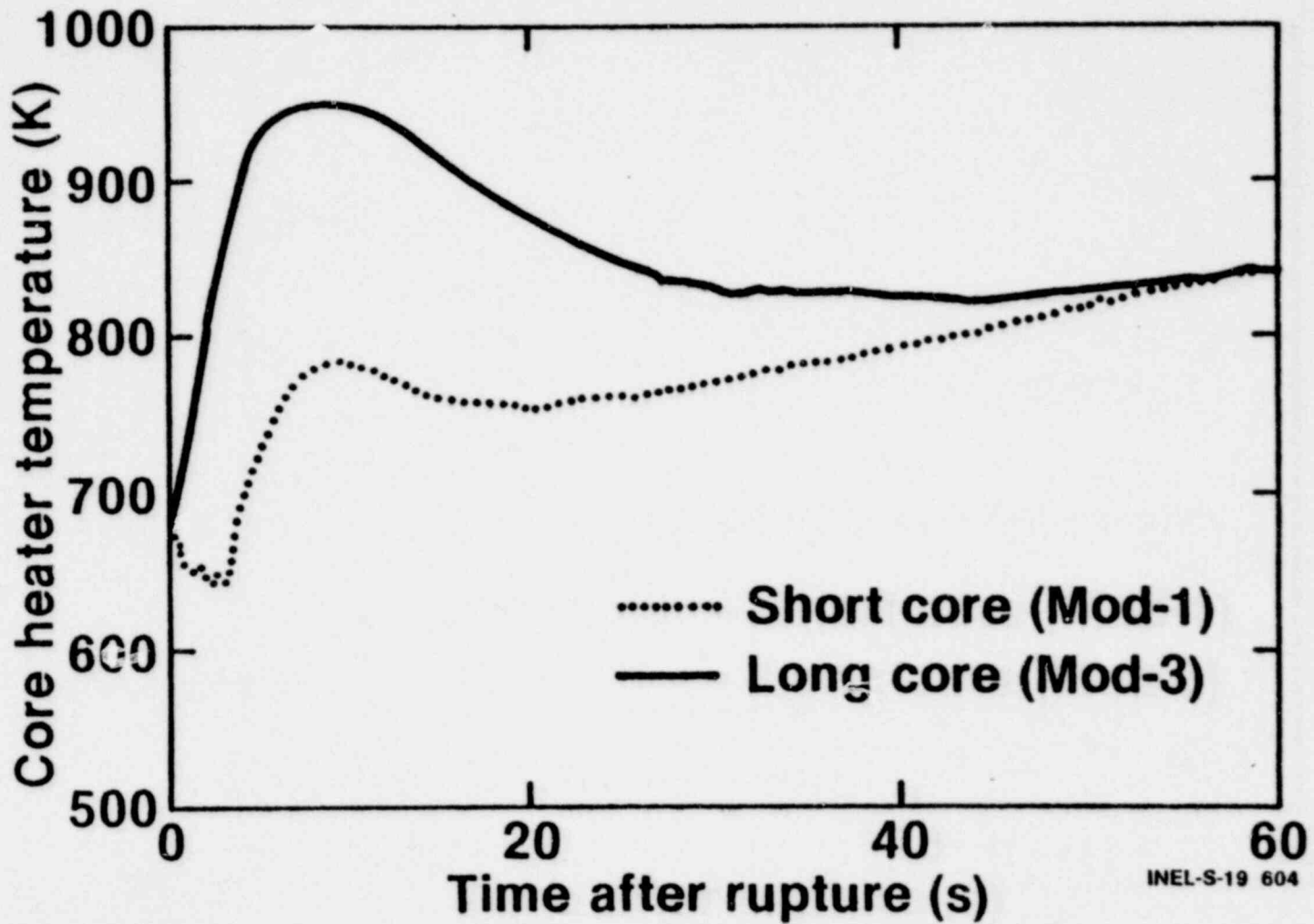


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Measured Rod Temperatures In Upper Core Regions of Semiscale



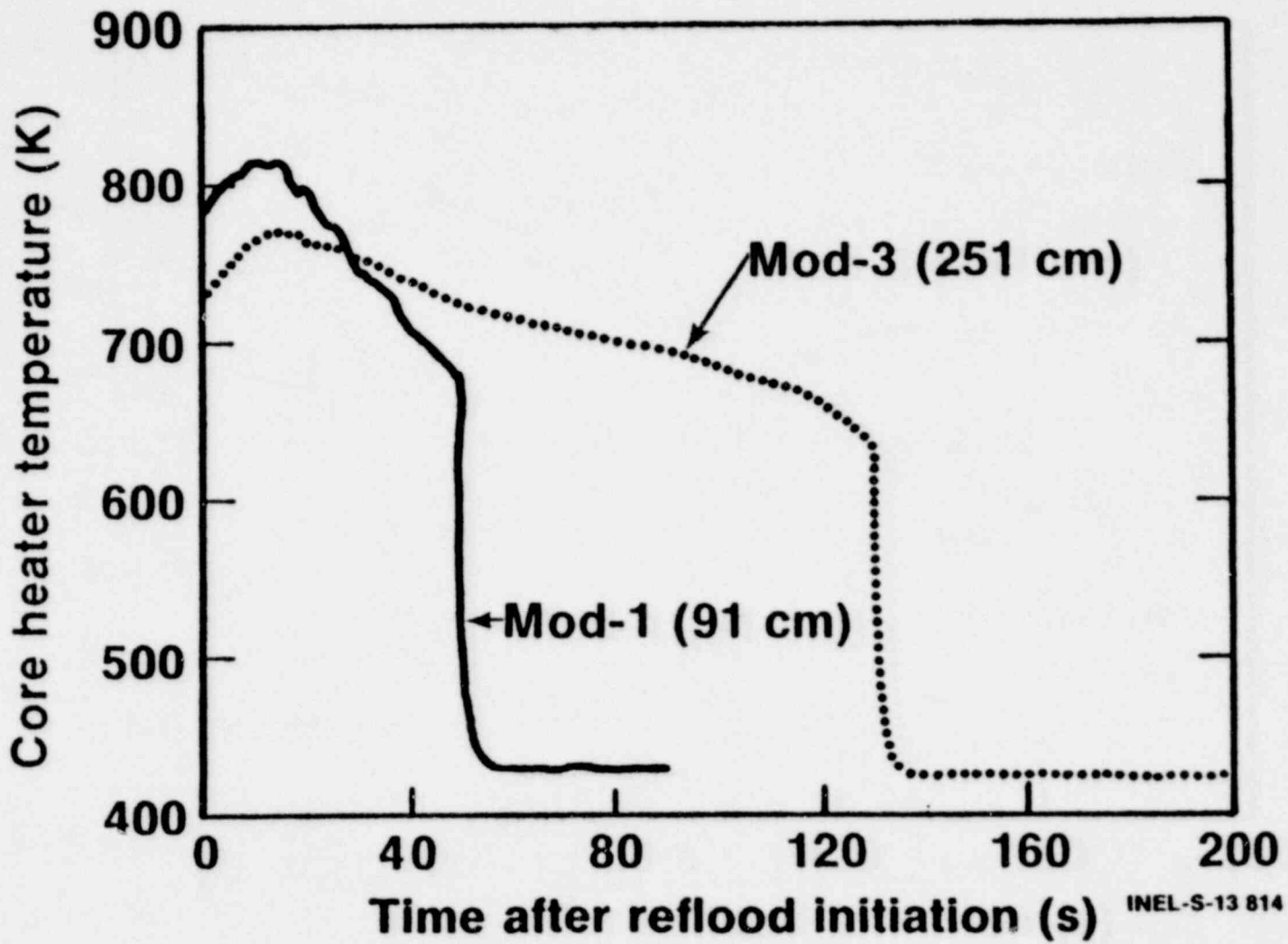
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Comparison of Cladding Temperatures

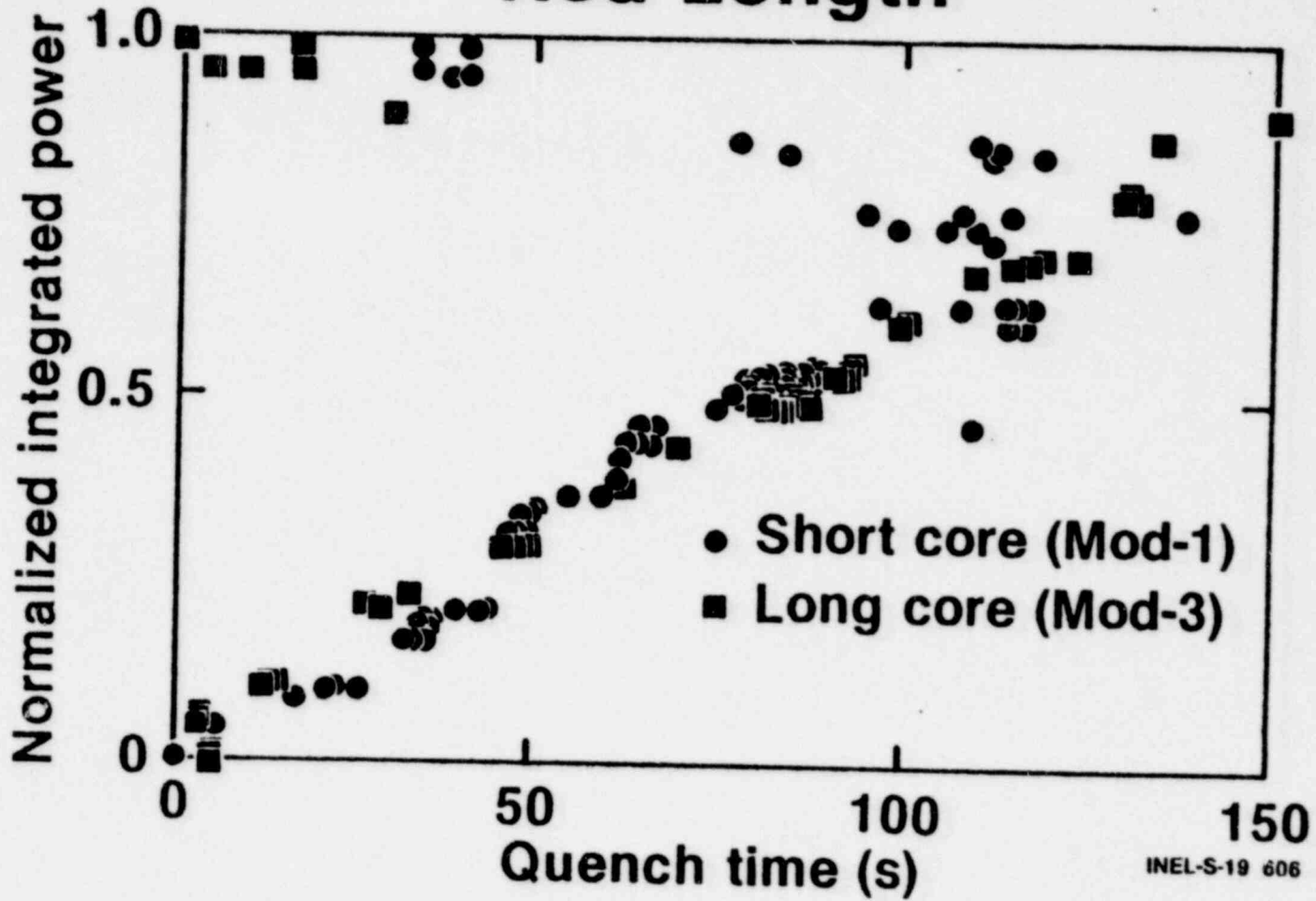


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INEL-S-13 814

Quench Times Corrected for Rod Length



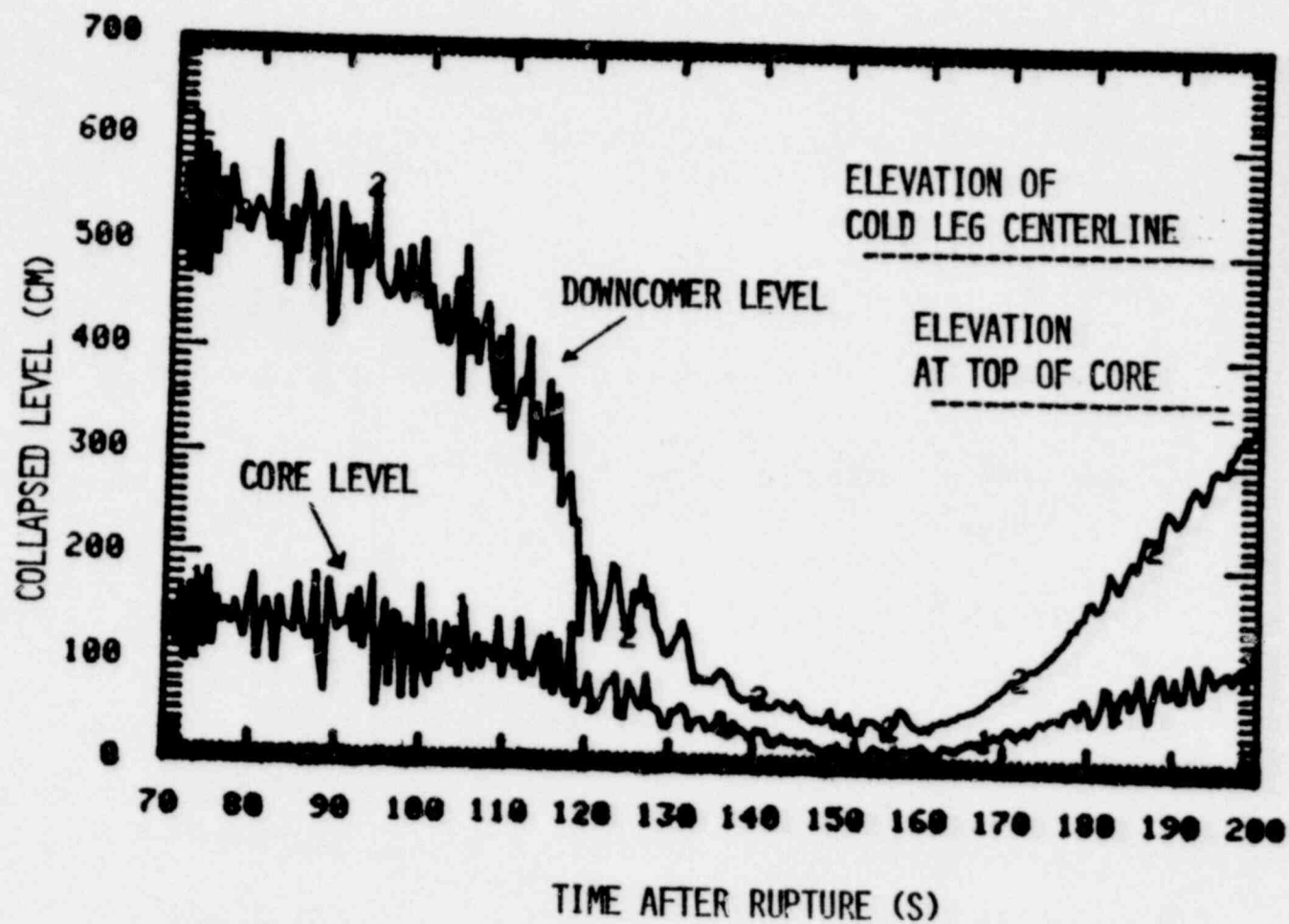
INEL-S-19 606

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DOWNCOMER AND CORE MASS DEPLETION COLLAPSED LIQUID LEVELS DURING TEST S-07-6

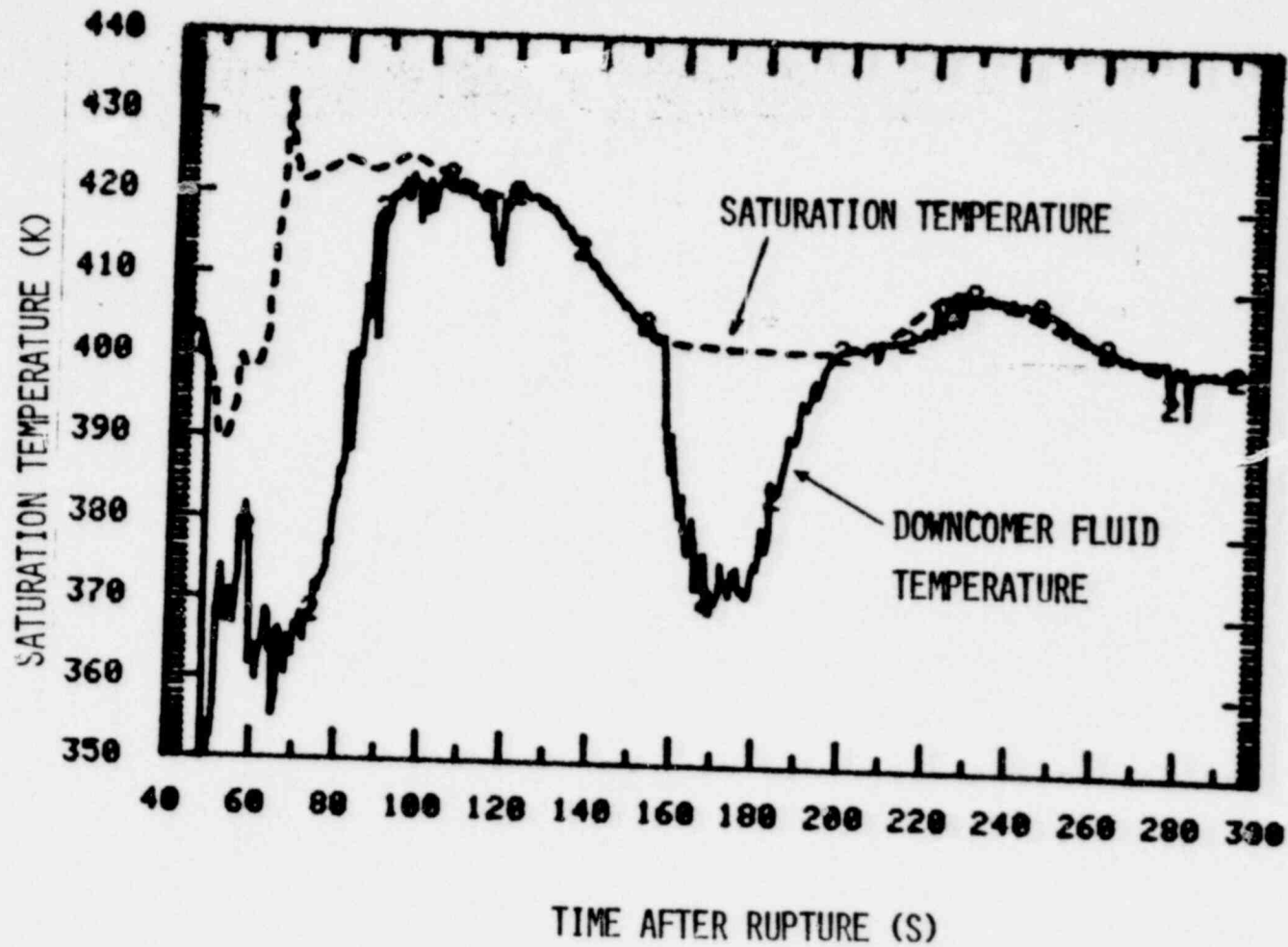


POOR ORIGINAL

POOR ORIGINAL

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COMPARISON OF SATURATION TEMPERATURE AND DOWNCOMER FLUID TEMPERATURE DURING MASS DEPLETION

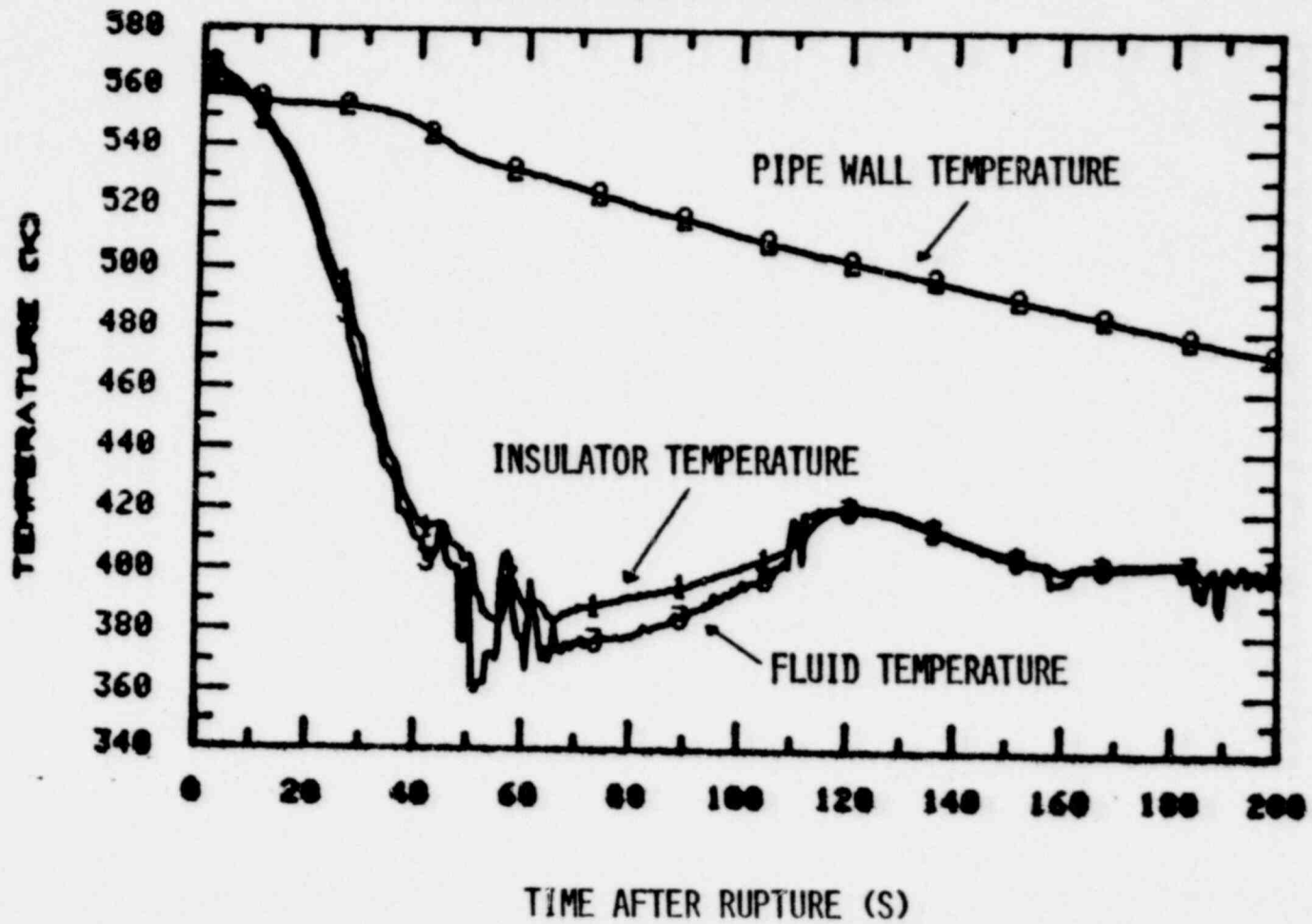


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POOR
ORIGINAL

COMPARISON OF PIPE WALL, INSULATOR, AND FLUID
TEMPERATURES IN DOWNCOMER

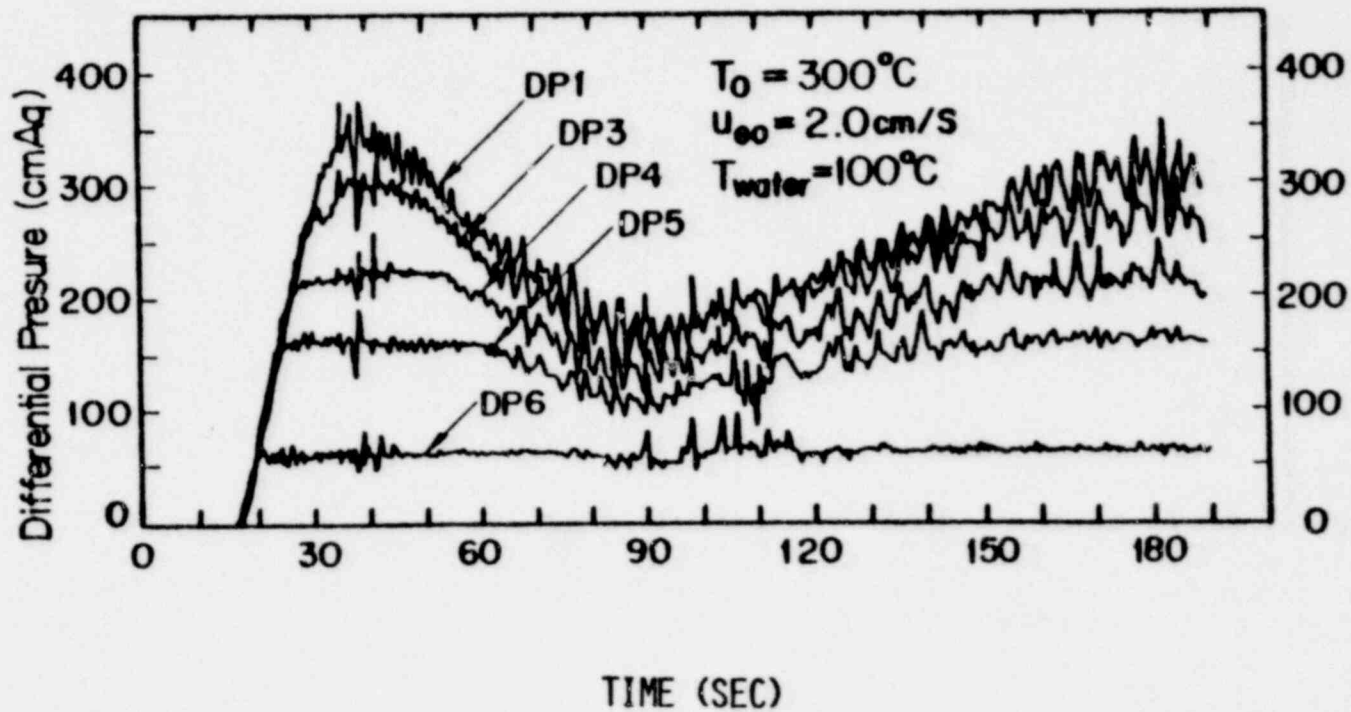


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JAPANESE DOWNCOMER EFFECTIVE WATER HEAD TEST RESULTS

DOWNCOMER HEAD DURING FILL AND DEPLETION



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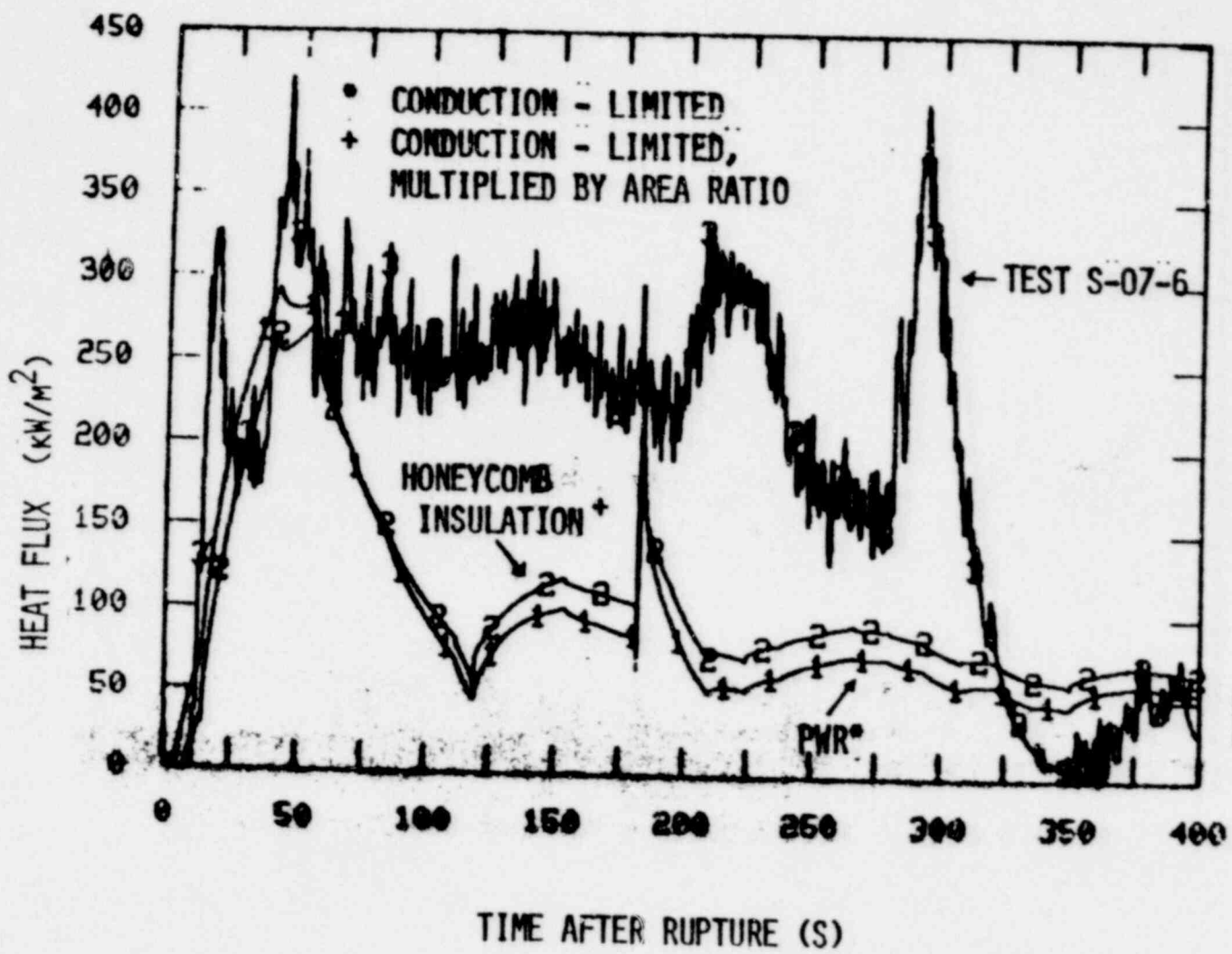
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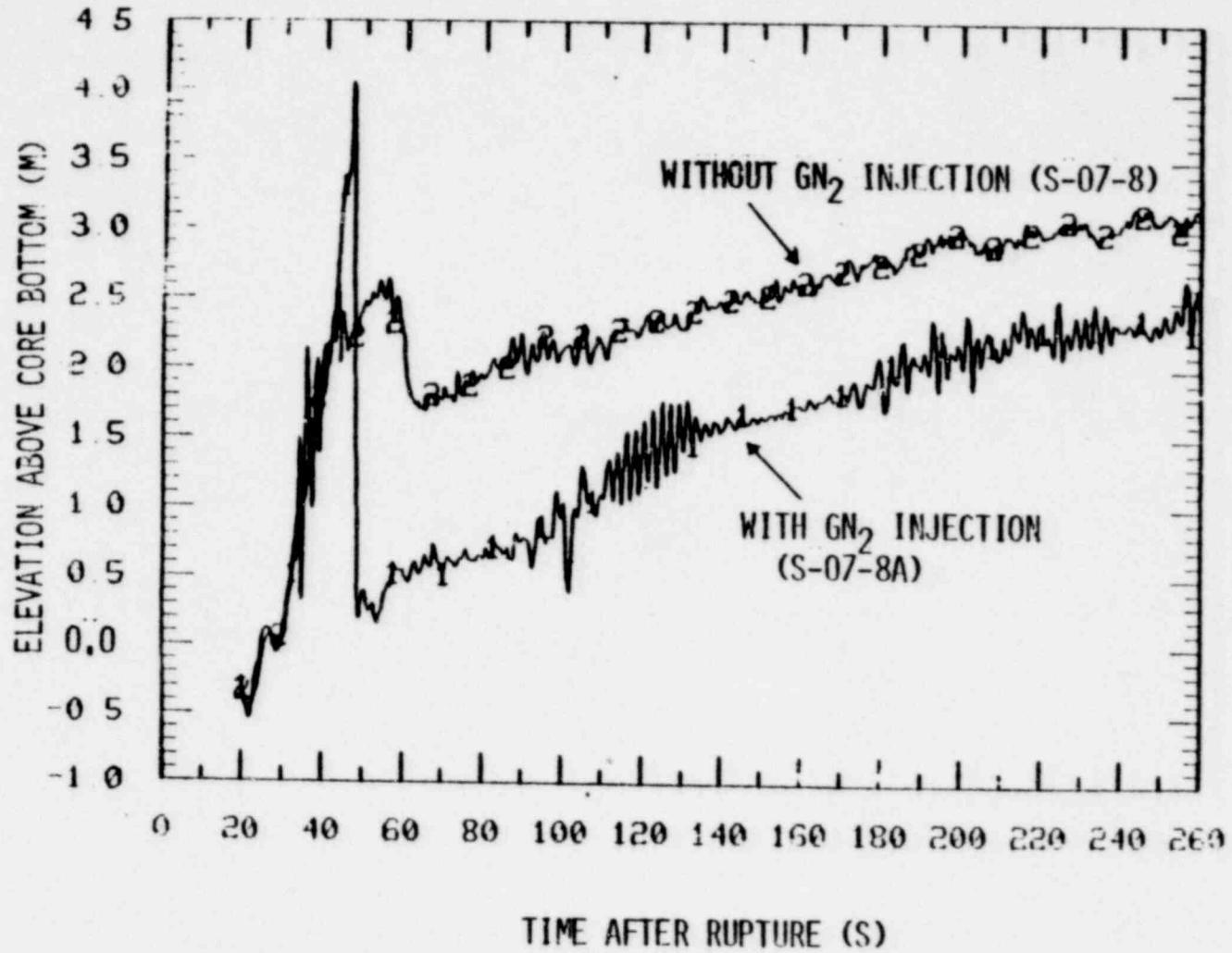
POOR ORIGINAL

COMPARISON OF DOWNCOMER WALL HEAT FLUXES



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COMPARISON OF CORE COLLAPSED LIQUID LEVEL WITH
AND WITHOUT ACCUMULATOR GN₂ INJECTION

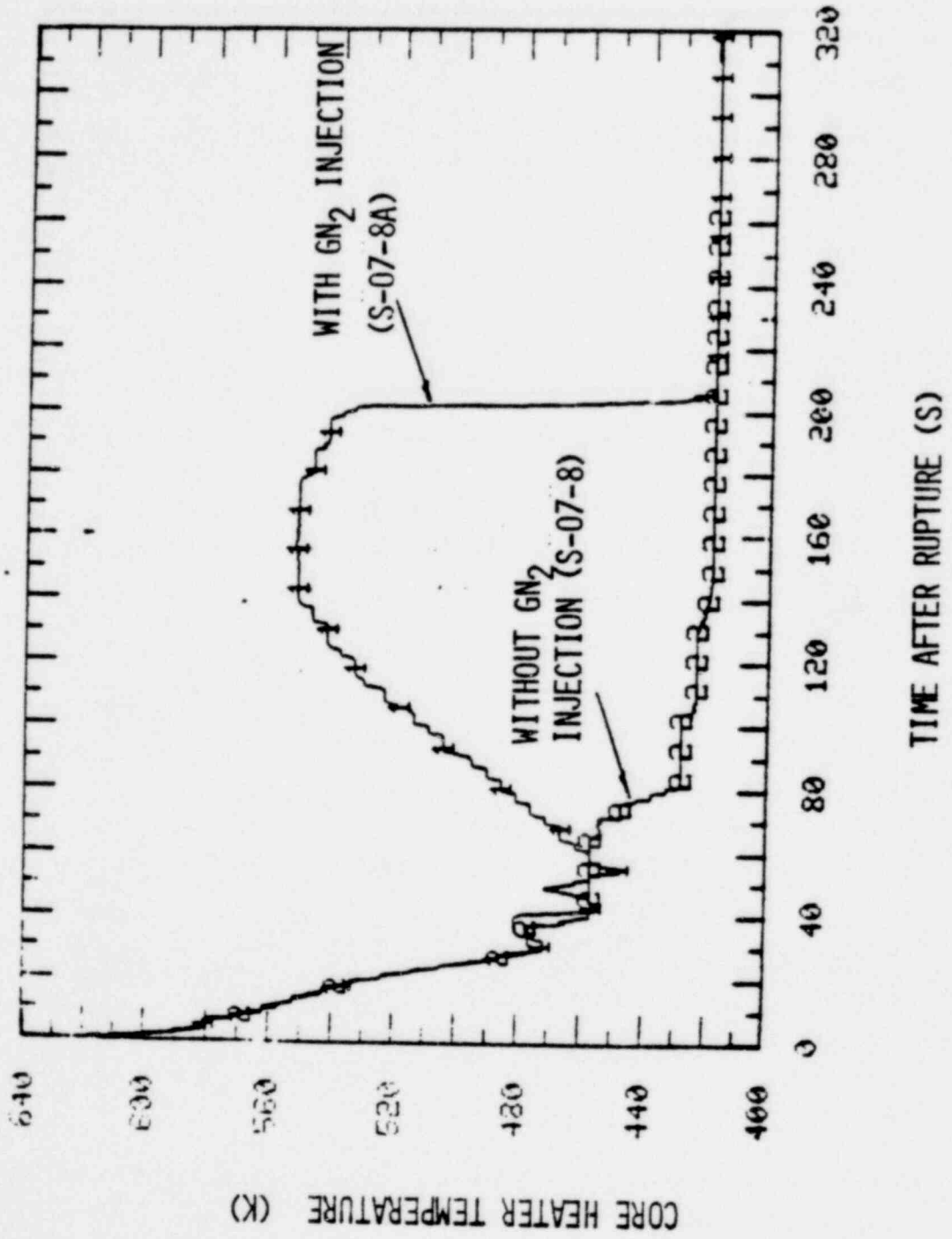


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COMPARISON OF UPPER CORE TEMPERATURE RESPONSE
WITH AND WITHOUT GN₂ INJECTION



POOR ORIGINAL

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TMI EXPERIMENTS

. NON-CONDENSIBLE BUBBLE TESTS

. TWO-HOUR SIMULATIONS

26

BUBBLE TEST RESULTS

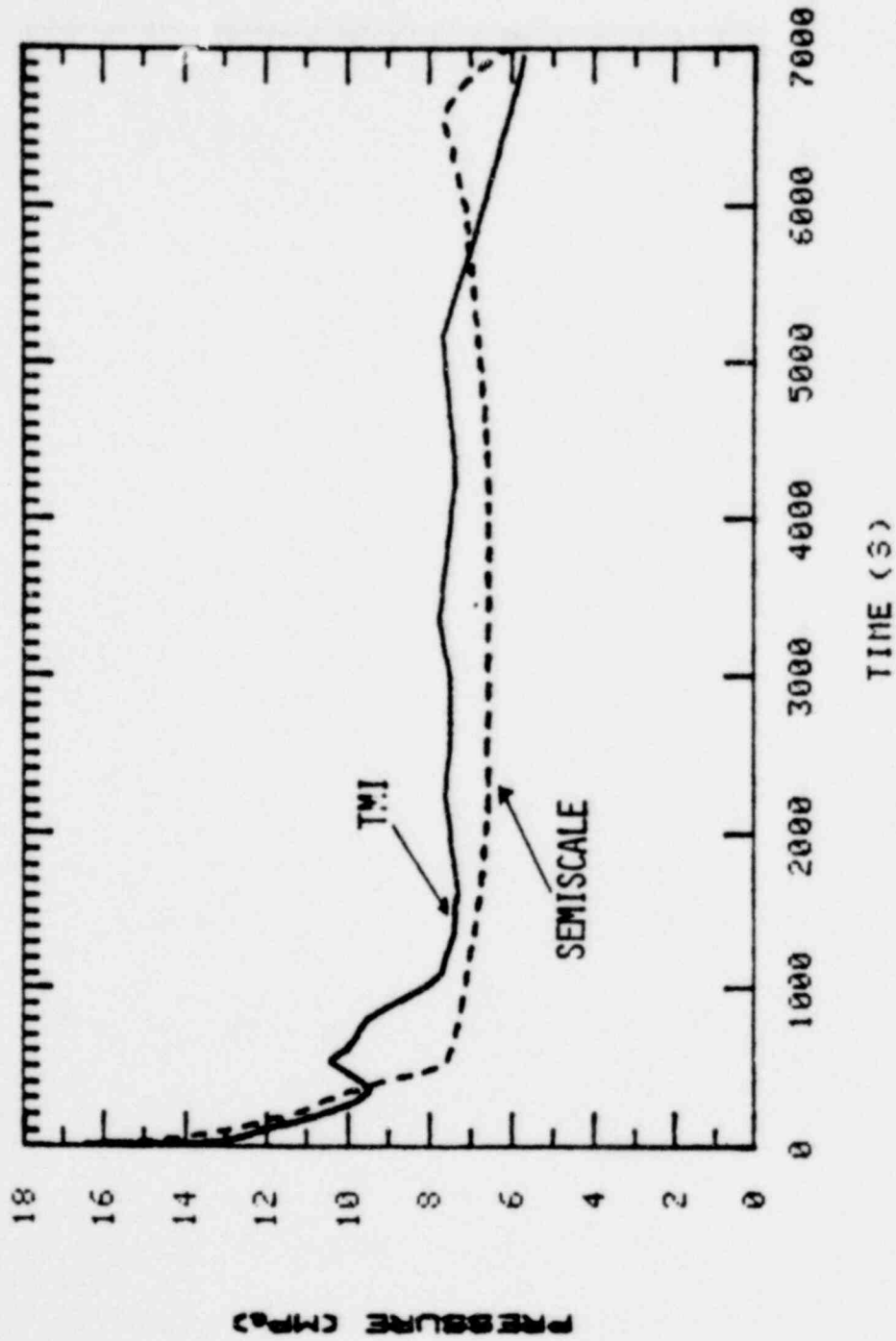
- . GAS TRANSPORT DURING DEPRESSURIZATION
- . MULTIPLE PUMP OPERATION

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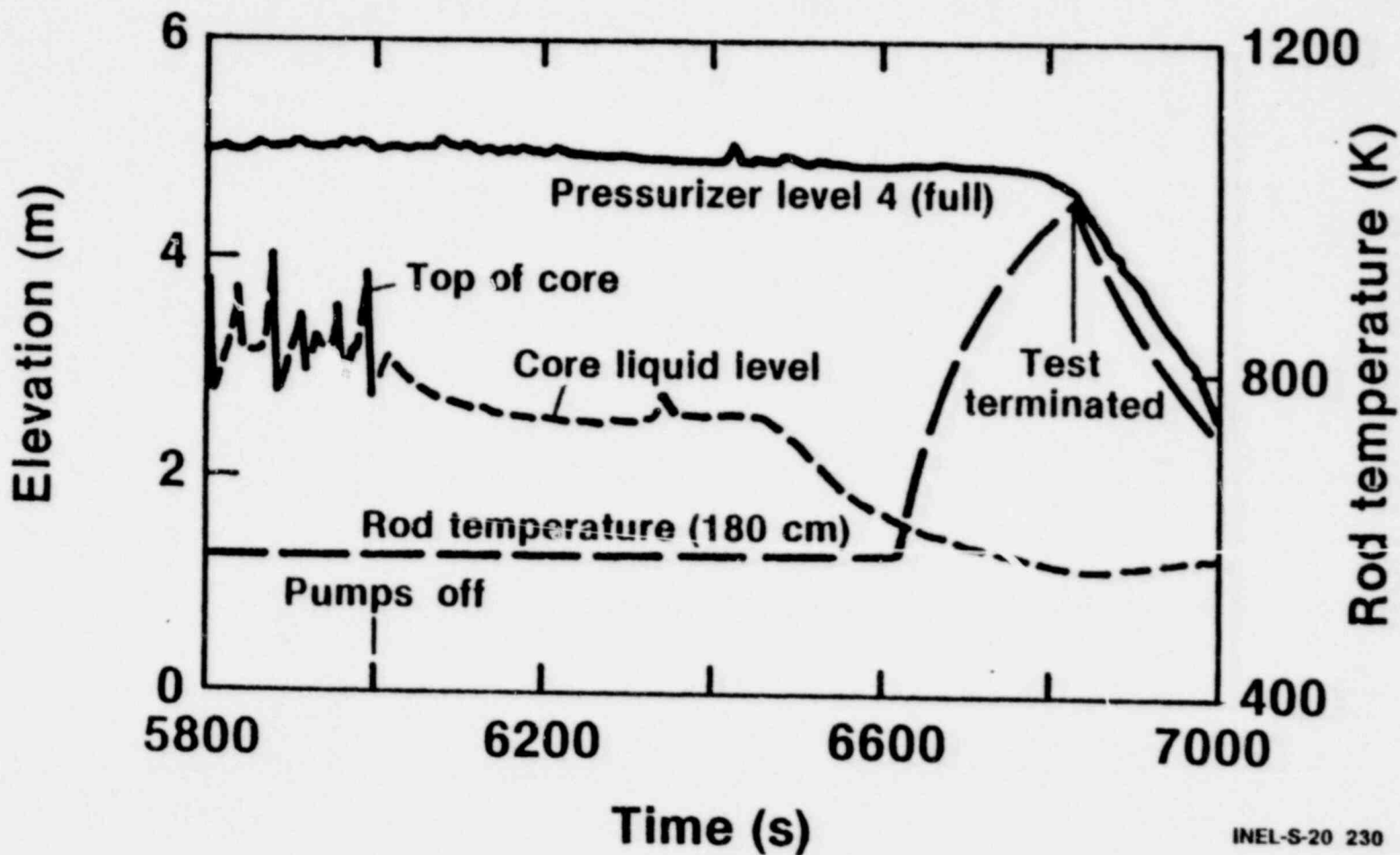
COMPARISON OF SYSTEM PRESSURE FOR THE TMI AND THE
SEMISCALE SIMULATION (S-TMI-31)



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Comparison of Collapsed Liquid Levels with Rod Temperature (Semiscale Test TMI-3I)

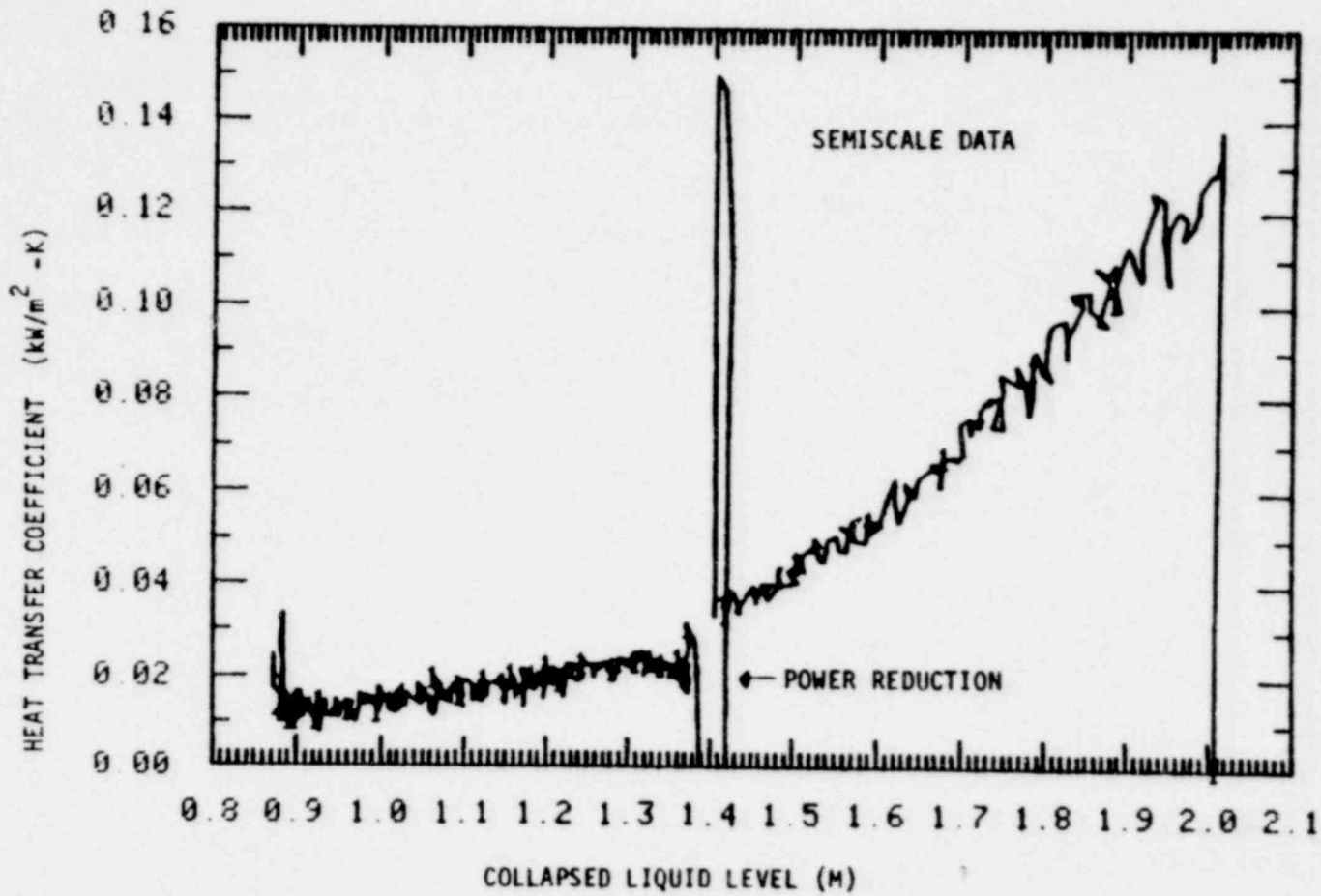


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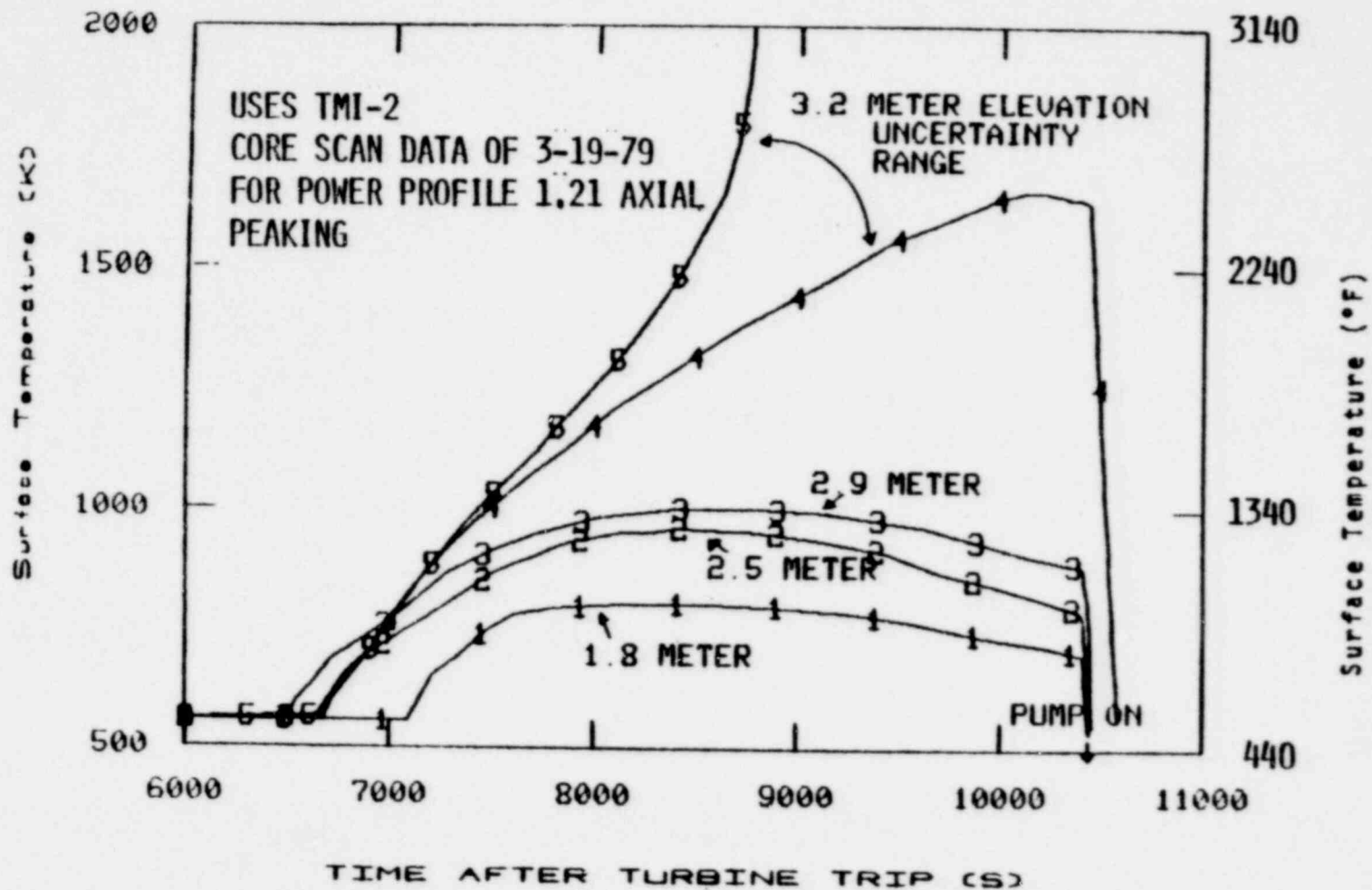
SEMISCALE HEAT TRANSFER COEFFICIENT VERSUS LIQUID LEVEL (TEST S-TMI-3C)



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ESTIMATED THREE MILE ISLAND CORE THERMAL RESPONSE



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NATURAL CIRCULATION AND UPPER HEAD DRAIN TESTS

- . FLOW BEHAVIOR
- . MEASUREMENT CAPABILITY
- . CODE BENCHMARK

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SUMMARY - RESULTS AND SIGNIFICANCE

- . LOFT DATA ENHANCED
 - CORE LENGTH
 - INACTIVE COMPONENTS

- . POTENTIALLY IMPORTANT PHENOMENA UNCOVERED
 - MASS DEPLETION
 - LOWER PLENUM INJECTION.

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Olson
8/27/79

SEMISCALE PROGRAM PLAN
FY-1980 - 1981

D. J. OLSON
ACRS MEETING
AUGUST 27, 1979



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SSS 1001

MOD-3 TEST SCHEDULE

FY-1978 AND FY-1979

- MOD-3 BASELINE TESTS (JUNE - NOVEMBER 1978)
- MOD-3 UHI SENSITIVITY TESTS (DECEMBER 1978 - APRIL 1979)
- TWO-PIPE CHARACTERIZATION TESTS (APRIL - JULY 1979)
(TENTATIVE)
- MOD-3 TWO-PIPE SENSITIVITY TESTS (SEPTEMBER 1979 - FEBRUARY 1980)
(TENTATIVE)

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THREE MAJOR PERTUBATIONS

- TEST S-07-6 ANOMOLOUS BEHAVIOR
- UHI CALCULATIONAL CAPABILITY
- THREE MILE ISLAND

SEMISCALE SCOPE

(DECEMBER 1978)

- COMPLETE BASELINE SERIES (APRIL 1979)
- UHI DRAIN TESTS (MAY 1979)
- TEST S-07-6 REPEAT (JUNE 1979)
- SMALL BREAK TESTING (JULY - SEPTEMBER 1979)
- INTERIM TEST SERIES (SEPTEMBER 1979 - JANUARY 1980)
- UHI SENSITIVITY TESTS (JANUARY - MAY 1980)

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SEMISCALE SCOPE

(AUGUST 1979)

- SMALL BREAK TESTING (AUGUST 1979 - MARCH 1980)
- TEST S-07-6 REPEAT (NOVEMBER 1979)
- SCOPING LOSS-OF-FEEDWATER TESTS (APRIL - OCTOBER 1980)
- UHI SENSITIVITY TESTS (NOVEMBER 1980 - MARCH 1981)
- CLOSED LOOP SECONDARY (MARCH - JUNE 1981)
- 2 X 4 LOOP SIMULATION (JULY 1981)

KEY PROGRAM REQUIREMENTS

- o PERFORM COMPARIBLE SMALL BREAKS TO LOFT AS SOON AS POSSIBLE

- o INSTALL NEW VESSEL INSULATORS, EXTERNAL INSULATION, NEW PWR SCALED STEAM GENERATOR PRIOR TO LOSS-OF-FEEDWATER TESTING

- o PRIOR TO CONVERSION TO OTHER THAN FOUR-LOOP CONFIGURATION A REEVALUATION OF PRIORITIES IS REQUIRED
 - UHI

 - TWO-PIPE DOWNCOMER

 - FOUR-LOOP OPERATIONAL TRANSIENTS

Harvego
8/28/79

FUTURE SEMISCALE TEST
PLANS AND SCHEDULE

E. A. HARVEGO
ACRS MEETING
AUGUST 27, 1979



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PROPOSED FUTURE TESTS

- SMALL BREAKS
- LOSS-OF-FEEDWATER
- UNDEFINED OPERATIONAL TRANSIENTS

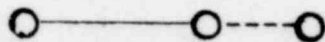
FUTURE SEMISCALE TESTS

1979

1980

1981

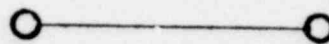
SMALL BREAK SERIES



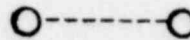
TEST S-07-6 REPEAT



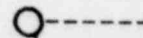
LOSS-OF-FEEDWATER SERIES



UHI SERIES



OPERATIONAL TRANSIENTS



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NEED FOR SMALL BREAK TESTS

- RECENT EVENTS
- LIMITED DATA
- CODE BENCHMARK DATA

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SEMISCALE SMALL BREAK TEST OBJECTIVES

- IDENTIFY IMPORTANT THERMAL-HYDRAULIC BEHAVIOR
- COMPARE WITH LOFT / AUDIT CALCULATIONS
- CODE EVALUATION

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PROPOSED SMALL BREAK TESTS

<u>TEST</u>	<u>BREAK SIZE</u>	<u>INITIAL CONDITIONS</u>	<u>CONFIGURATION</u>
S-SB-2	2.5%	PWR AUDIT CALCULATIONS	MOD-3
S-SB-4	2.5%	LOFT TEST L3-1	LOFT
S-SB-3	2.5%	PWR AUDIT CALCULATIONS	LOFT
S-SB-5	0.16%	PWR AUDIT CALCULATIONS	MOD-3
S-SB-1	10%	TEST S-07-10	MOD-3 (UHI)

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EXPECTED..SMALL..BREAK..RESULTS

- CORE UNCOVERY
- NATURAL CIRCULATION PHENOMENA
- SECONDARY SIDE INFLUENCES
- SCALING EFFECTS (LOFT/SEMISCALE)

LOSS-OF-FEEDWATER RATIONALE

- RELATIVE PROBABILITY
- EVENT TREE WELL DEFINED
- LACK OF EXPERIMENTAL DATA

LOSS-OF-FEEDWATER OBJECTIVES

- SYSTEM RESPONSE
- MULTIPLE FAILURES
- CODE DEVELOPMENT

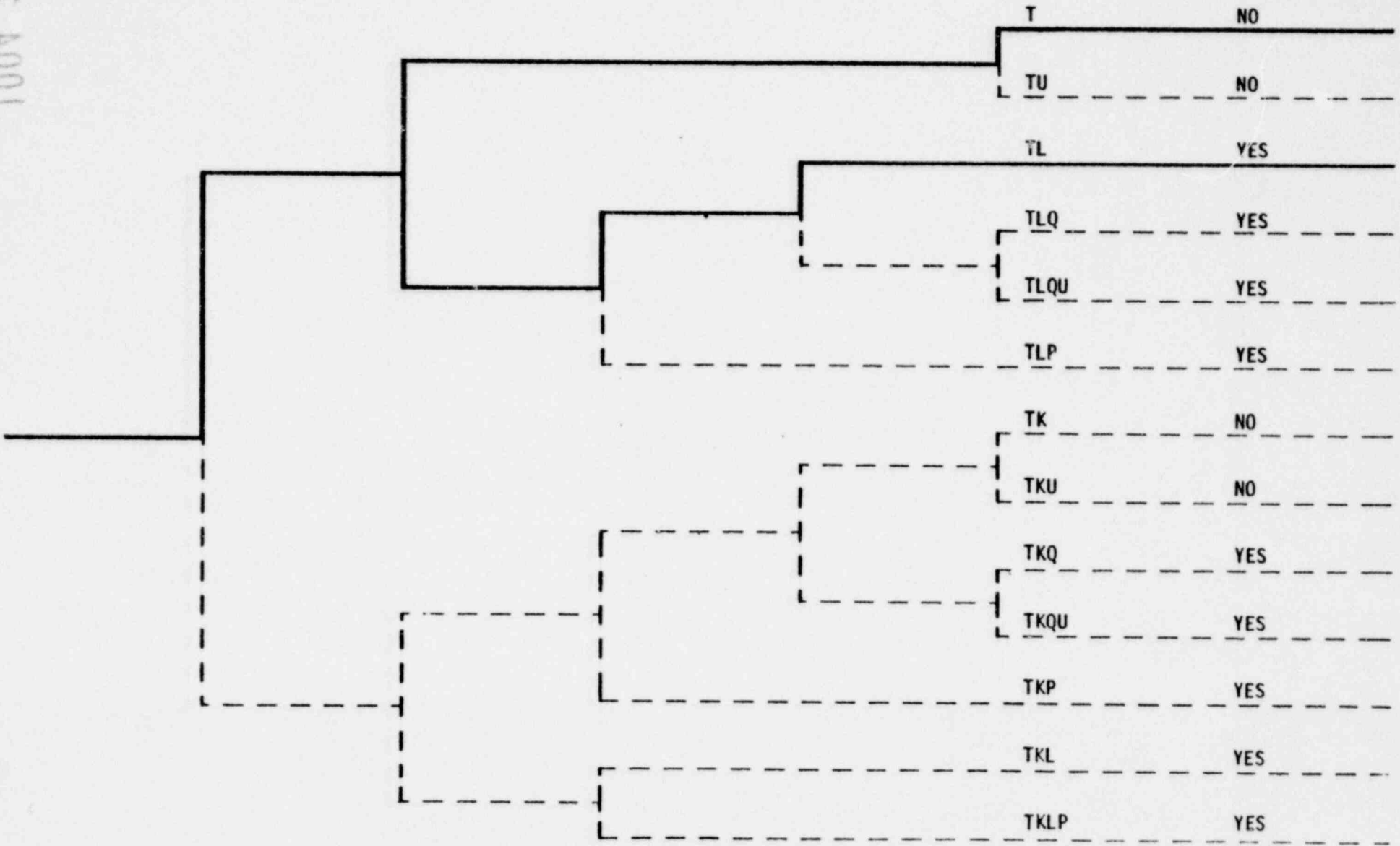
LOSS-OF-FEEDWATER TEST PLAN

- LOSS-OF-FEEDWATER INITIATES TEST
- EVENT TREE APPROACH
- TEST SERIES PROCEEDS FROM INITIATING EVENT TO LIMITING CASE

EVENT TREE FOR LOSS-OF-FEEDWATER TRANSIENT

INITIATING EVENT LOSS-OF-FEEDWATER	REACTOR PROTECTION SYSTEM (SCRAM)	SEC. STEAM RELIEF AND AUXILIARY FEEDWATER	PRESSURIZER RELIEF VALVE OPENS	PRESSURIZER RELIEF VALVE RECLOSSES	HIGH PRESSURE INJECTION	CORE DAMAGE
T	K	L	P	Q	U	

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EXPECTED LOSS-OF-FEEDWATER RESULTS

- IDENTIFY IMPORTANT THERMAL-HYDRAULIC EFFECTS
- IDENTIFY LARGE REACTOR MEASUREMENT CAPABILITIES / LIMITATIONS
- ASSESS CODE ABILITY TO PREDICT OUTCOME OF EXPERIMENTAL PATH

POSSIBLE FUTURE TRANSIENTS

- INCREASED SECONDARY HEAT REMOVAL (SHORT TERM)
- DECREASED SECONDARY HEAT REMOVAL (SHORT TERM)
- DECREASED REACTOR COOLANT SYSTEM FLOW (SHORT TERM)
- REACTIVITY AND POWER DISTRIBUTION ANOMALIES (MEDIUM / LONG TERM)
- INCREASE IN REACTOR COOLANT INVENTORY (MEDIUM / LONG TERM)
- DECREASE IN REACTOR COOLANT INVENTORY (MEDIUM / LONG TERM)
- ANTICIPATED TRANSIENT WITHOUT SCRAM (LONG TERM)

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OBJECTIVES OF OPERATIONAL TRANSIENTS

- SYSTEM RESPONSE TO SINGLE INITIATING EVENT
- INFLUENCES OF MULTIPLE FAILURES
- LARGE REACTOR INSTRUMENTATION CAPABILITIES / LIMITATIONS
- DATA BASE FOR CODE DEVELOPMENT

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TEST PLAN FOR OPERATIONAL TRANSIENTS

- EACH INITIATING EVENT BASIS FOR TEST SERIES
- TEST SERIES USES EVENT TREE APPROACH
- SERIES PROCEEDS FROM INITIATING EVENT TO LIMITING CASE
- FOLLOWUP TESTING DEVELOPED FROM EACH TEST SERIES
(SYSTEM RECOVERY, OPERATOR INTERVENTION)

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EXPECTED OPERATIONAL TRANSIENT RESULTS

- IDENTIFY POTENTIAL THERMAL-HYDRAULIC BEHAVIOR FOR A WIDE RANGE OF OPERATIONAL TRANSIENTS
- IDENTIFY POTENTIAL OPERATOR ACTIONS TO MITIGATE CONSEQUENCES
- IDENTIFY CODE ABILITY TO PREDICT LIMITING CASES AND ASSESS OPERATING MARGINS
- BASIS FOR DEFINING SEPARATE EFFECTS EXPERIMENTS

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Supplement -
Johansen
8/28/79

1004 245

SEMISCALE PROGRAM ACCOMPLISHMENTS

SUPPLEMENTARY SLIDES

OBJECTIVES:

- o INVESTIGATE NONCONDENSIBLE GAS BUBBLE BEHAVIOR DURING PRESSURIZER VENTING AND QUALIFY SYSTEM RECOVERY PROCEDURES

- o ESTABLISH SYSTEM THERMAL-HYDRAULIC RESPONSE TO TMI SEQUENCE OF EVENTS
 - QUANTIFY PRESSURIZER LIQUID LEVEL BEHAVIOR
 - IDENTIFY SEMISCALE HARDWARE AND CONFIGURATION LIMITATIONS WITH REGARD TO EXTREMELY SMALL BREAK SIMULATIONS

HARDWARE CHANGES MADE TO SEMISCALE SYSTEM FOR TMI SIMULATIONS

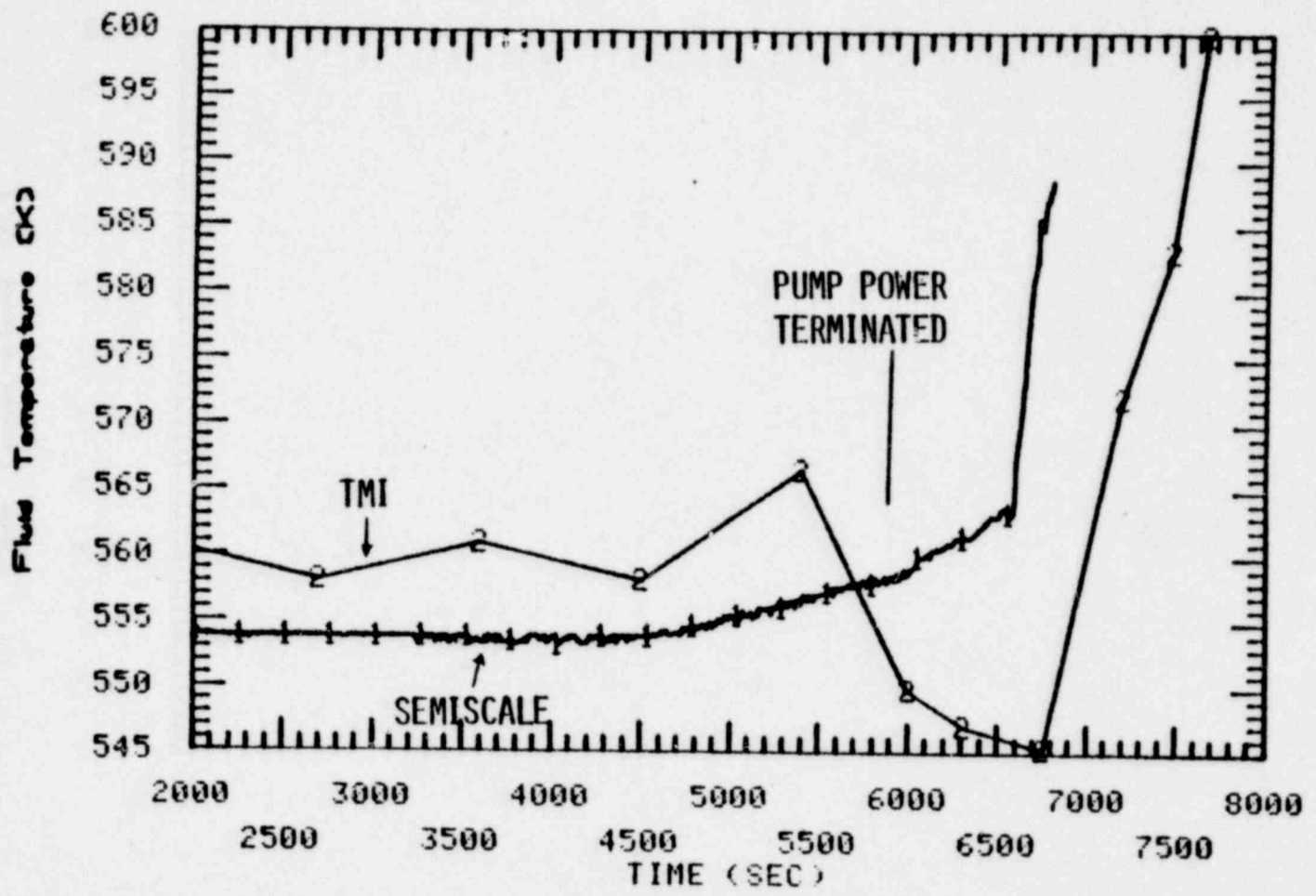
- o PRESSURIZER SURGE LINE ELEVATION AND RESISTANCE MODIFIED
- o ADDITION OF PRESSURIZER SAFETY VALVES AND POWER OPERATED VALVE
- o ADDITION OF UPPER PLENUM TO INLET ANNULUS VENT LINE AND VALVE
- o REDUCED LOOP RESISTANCE
- o ADDED ORIFICE IN HOT LEG VESSEL NOZZLE TO PROMOTE SYMMETRIC LOOP RESPONSE

SEMISCALE SYSTEM OPERATING LIMITATIONS

- o EXTERNAL HEAT LOSS
- o STEAM GENERATOR DESIGN
- o PCP DEGRADATION

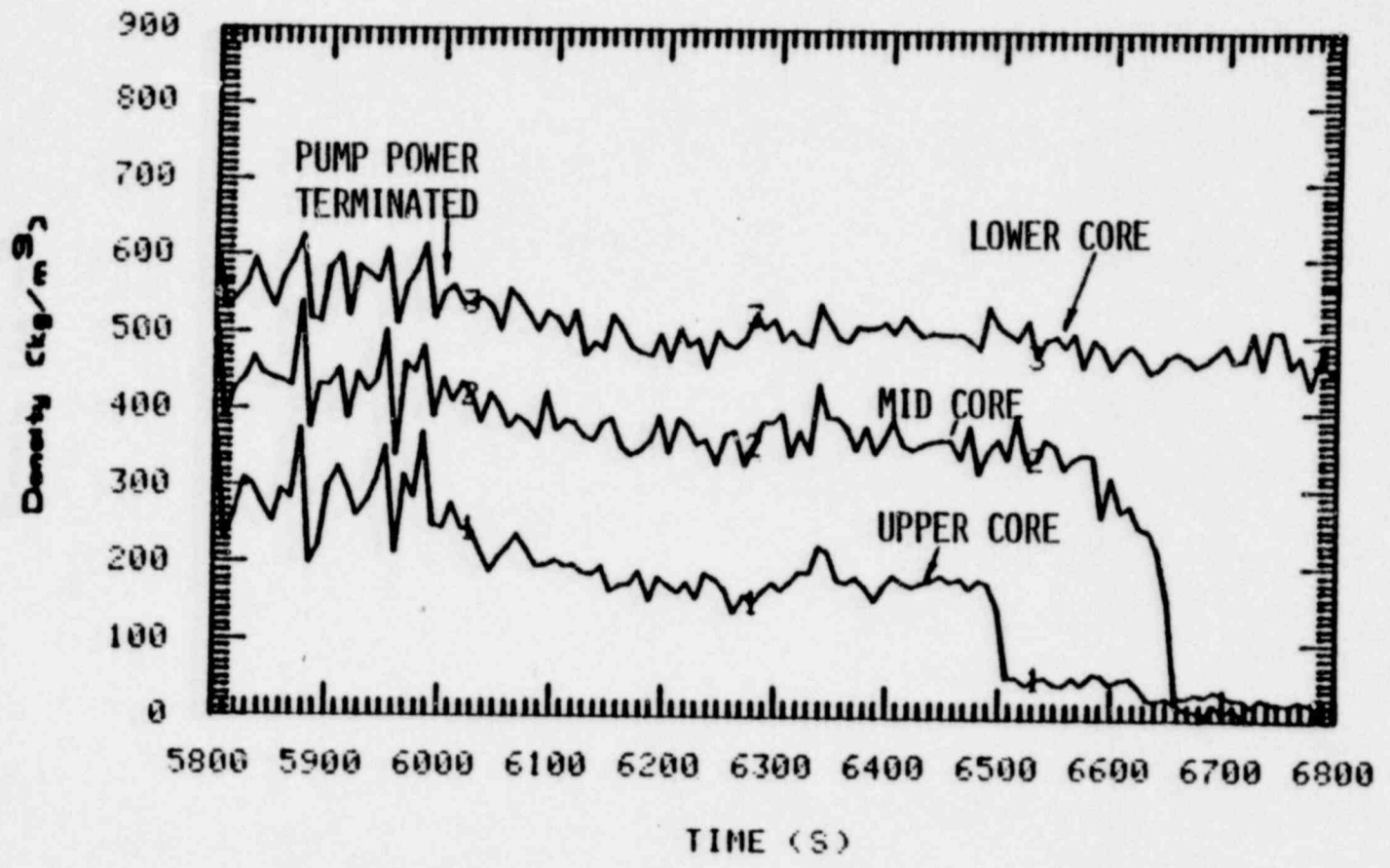
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SEMISCALE VERSUS TMI HOT LEG TEMPERATURE



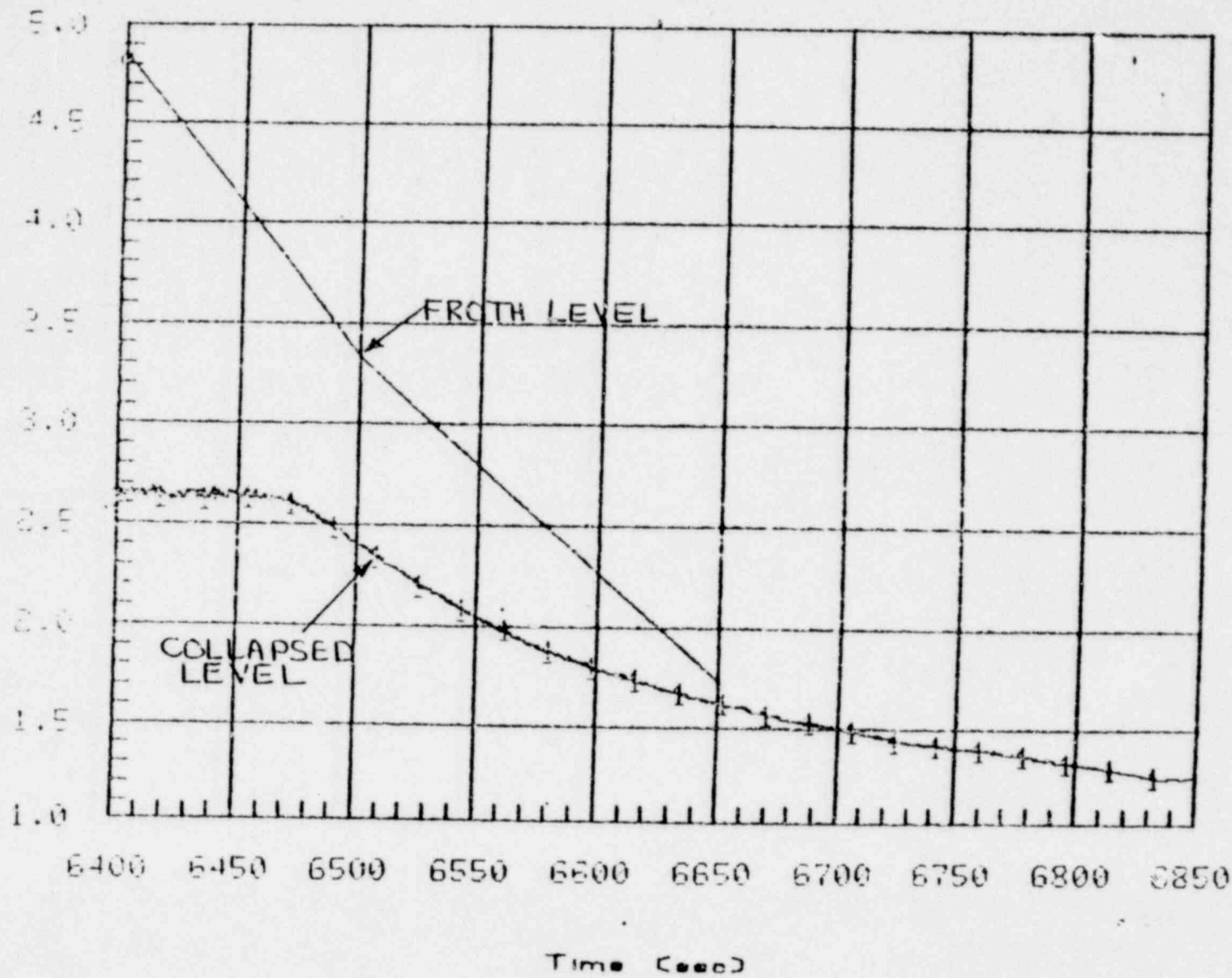
POOR ORIGINAL

CORE FLUID DENSITY DISTRIBUTION FOR TEST TMI-31



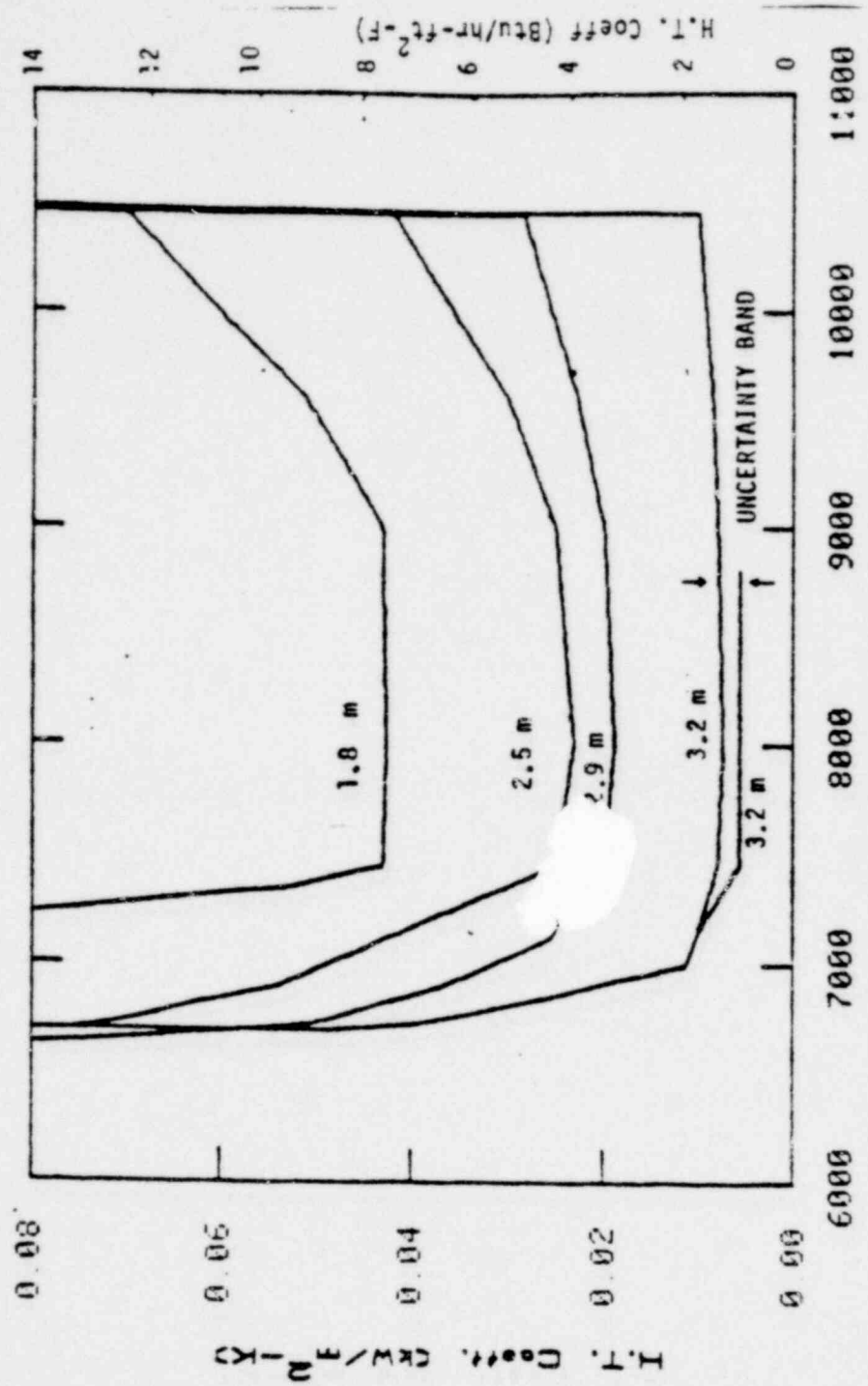
POOR
ORIGINAL

COMPARISON OF CORE FROTH AND COLLAPSED LIQUID LEVELS FOR TEST TMI-31



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HEAT TRANSFER COEFFICIENTS ESTIMATED FOR TMI BASED ON TSAT



TIME AFTER TURBINE TRIP (S)

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KEY POINTS

- OVERALL THERMAL-HYDRAULIC TRENDS OBSERVED IN THE SEMISCALE SIMULATIONS WERE SIMILAR TO THOSE OBSERVED IN THE TMI DATA
- SEMISCALE RESULTS TEND TO INDICATE THAT THE TMI PRESSURIZER LEVEL INDICATION WAS VALID FOR AT LEAST THE FIRST FEW HOURS OF THE TRANSIENT, BUT WAS AN INAPPROPRIATE INDICATION OF SYSTEM MASS INVENTORY
- SEMISCALE CORE RESPONSE AFTER PUMP SHUTDOWN INDICATES VERY POOR HEAT TRANSFER ON EXPOSED ROD LOCATIONS
- SEMISCALE RESULTS SUGGEST THAT SIGNIFICANT DAMAGE COULD HAVE OCCURRED IN THE UPPER SECTION OF THE TMI CORE
- SCALING DISTORTIONS MUST BE CONSIDERED WHEN CONDUCTING SLOW TRANSIENTS IN THE SEMISCALE FACILITY

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