

ROUGH DRAFT

SAND79-1963C

TWO-PHASE JET LOADS

David Tomasko
Sandia Laboratories
Albuquerque, NM 87185

Presented at the Seventh Water Reactor Safety Research Information Meeting, November 5-9, 1979, Gaithersburg, Maryland.

1602 321

Two-Phase Jet Loads*

David Tomasko
Sandia Laboratories
Albuquerque, NM 87185

Prediction of Two-Phase Jet Forces

The ultimate objective of the Sandia Labs Program on two-phase jet loads is the development of an approximate engineering model to characterize two-phase jets emanating from circumferential or longitudinal breaks in a typical PWP piping system.

To accomplish this objective two concurrent paths will be followed. The first is to develop a model analytically beginning with first principles and making the minimum number of approximations. The Moody^{6,7} model, currently used by NPC licensing, will serve as a starting point for this development.

The second path is the use of large computer codes to produce data on a wide variety of full scale PWR pipe breaks. These data will form the structure of an engineering model which will supplement and support the analytic expression and, in addition, provide more specific and accurate predictions in certain cases. To increase the reliability of these data, the codes will be verified against existing two-phase jet load data from small scale up to approximately full scale.

Most of the work performed to date has focused on this second task. Three computer codes have been employed to investigate jet load experiments in detail. These are: BEACON/MOD2-INFL⁽¹⁾; CSQ - Sandia Labs⁽²⁾; and TPAC-PLA - LASL⁽³⁾. Experimental data used in this evalaution were obtained from two sources in the

*This work was supported by the United States Nuclear Regulatory Commission.

Federal Republic of Germany: PS-93⁽⁴⁾ - Kraftwerk Union, Erlangen; and PS-50⁽⁵⁾ - Battelle-Frankfurt, Frankfurt. Based on comparisons with data and code evaluation, the following conclusions have been drawn:

BEACON/MOD2 - BEACON/MOD2 exhibited a high degree of sensitivity to the evaporation rate multiplier (λ_e) and the drag coefficient (K) used in the constitutive relations. It had to be run as a containment code, requiring time dependent input boundary conditions which are generally not experimentally available, and it required the LAFL equation of state (EOS) option (incompressible liquid and ideal gas steam). These problems precluded its use in a detailed two-phase jet parameter study. However, many of these limitations may be overcome in the MOD3 version to be released later this year. At that time, BEACON will be re-evaluated.

CSC - CSC did a fairly good job of simulating a two-phase jet under equilibrium conditions. Its multi-material capability allowed the determination that the jet-air interaction is probably small and could be ignored in future work. It seems, however, that early time non-equilibrium phenomena may be important in two-phase jet modelling. This necessitates using a code with a non-equilibrium capability.

TRAC-PLA - Of the codes evaluated to date, TPAC-PLA produced the best agreement with the jet data. TRAC-PLA was employed in two distinct modes. A 1-D pipe representation

for break flow and a vessel representation. The 1-D pipe simulation consisted of an accumulator and pipe connected to a break. Initially the pipe and accumulator were set to conditions matching those of the experiment being simulated, while the break was maintained at ambient conditions. Since TRAC-PIA does not have a containment component model, a unique vessel model was developed to simulate fluid reservoir, piping, and containment. The vessel model used consisted of a vessel component which had all internals removed and appropriate flow areas set to zero to simulate the experimental blowdown geometry and connecting pipes exiting to breaks to allow venting of the blowdown fluid. The calculation of steady-state blowdown forces

$$F_B = (P_F - P_A)A + \rho v^2 A$$

and pipe reaction forces, $F_R = F_B + F_W$,

where
$$F_W = \frac{\partial}{\partial t} \int_0^L \rho v A dz$$

was possible with either model. Two-phase jet characteristics, however, could only be obtained with the vessel model.

Comparisons of TRAC-PIA pipe models to Kraftwerk Union and Battelle-Frankfurt data generally yield good agreement. They also pointed out the need to accurately model friction effects during blowdown.

Early attempts at using the vessel models failed because of large pressure spikes and asymmetries which appeared to be caused by water packing in downstream cells. This problem was greatly reduced by a modification of the time-step control logic. If $\Delta P/P$ for any cell exceeded a maximum value (1%) for any time step, that time step was reduced by a factor of two, and if $\Delta P/P$ exceeded a smaller value (3%) time step growth was inhibited. Results obtained with vessel models using the modified time step logic are generally in fairly good agreement with experimental data. They also point out the need of using non-equilibrium thermodynamics during the initial transient portion of the blowdown.

Present work on two-phase jets includes performing a parametric study of typical PWR cold leg breaks (ZION) using TRAC-P1A, modelling other tests and facilities with TPAC-P1A and CSQ (HDP, CANON, MARVIKEN), obtaining additional experimental data for model assessment, evaluating other computer codes (KFIX, COMMIX-2, and PEACON/MOD3 when available), and beginning the development of an approximate engineering model.

1602 325

PB 80 8001

References:

1. C. P. Proodus, S. W. James, W. H. Lee, J. F. Lime, and P. A. Pate "BEACON/MOD2 A CDC 7600 Computer Program for Analyzing the Flow of Mixed Air, Steam, and Water in a Containment System," CDAP-TR-002, EG&G Idaho, Dec. 1977.
2. S. L. Thompson, "CSQ-A Two-Dimensional Hydrodynamic Program With Energy Flow and Material Strength," SAND74-0122, Sandia Laboratories, Albuquerque, NM 87185, Aug. 1975.
3. TPAC-P1A, "An Advanced Best Estimate Computer Program for PWR LOCA Analysis," NUREG/CR-0063, LA-7279MS Vol. 1.
4. R. Fichler, W. Kastner, F. Lochner, and F. Fiedle, "Studies on Critical Two-Phase Flow," NRC-477.
5. "LOCA Experiments with a PWR Multi-Compartment Model Containment," BF-RS50-62-5, Battelle Institut, F. V. Frankfurt Am Main.
6. "Fluid Reaction and Impingement Loads," F. J. Moody, Conference on Structural Design of Nuclear Plant Facilities, Vol. 1, Chicago, Illinois, Dec. 17-18, 1973.
7. "Prediction of Blowdown Thrust and Jet Forces," F. J. Moody, ASME Paper No. 69-HT-31, August, 1969.

1602 326

I. PURPOSE -- THE DEVELOPMENT OF AN IMPROVED APPROXIMATE ENGINEERING MODEL TO CHARACTERIZE TWO-PHASE JETS EMANATING FROM CIRCUMFERENTIAL OR LONGITUDINAL BREAKS IN A TYPICAL PWR PIPING SYSTEM.

1602 327

MODEL DEVELOPMENT STRATEGY

1. SUBJECT GROUNDWORK
2. CODE INSTALLATION -- BEACON/MOD2, TRAC
3. CODE INVESTIGATION -- BEACON/MOD2, TRAC, CSQ
4. DEVELOP BASIC NODALIZATION
5. PERFORM PARAMETER STUDY
6. PERFORM CODE ASSESSMENT USING EXPERIMENTAL DATA OBTAINED FROM THE OPEN LITERATURE AND FOREIGN SOURCES
7. DEVELOP AN APPROXIMATE ENGINEERING MODEL BASED ON CALCULATIONS, EXPERIMENTAL DATA AND PHYSICAL ARGUMENTS

1602 328

COMPUTER CODES FOR TWO-PHASE JETS

1. BEACON/MOD2

CAPABILITIES

- A. Two-Phase Two-Component
- B. UVUT
- C. 1-D, 2D, 2D-Axisymmetric
- D. Finite Difference with Eulerian Mesh
- E. Incompressible Fluid EOS

2. CSQ

CAPABILITIES

- A. Multi-Material
- B. TE
- C. 2-D Eulerian-Lagrangian Code
- D. Modelling by Eulerian Mesh (Perpendicular Cell Boundaries)

3. TRAC-P1A

CAPABILITIES

- A. Two-Phase
- B. UVUT
- C. 1-D Implicit Finite Difference for Pipes and Valves (Newton-Raphson)
- D. 3-D Implicit Finite Difference for Vessel (Gauss-Seidel)
- E. Model by Component

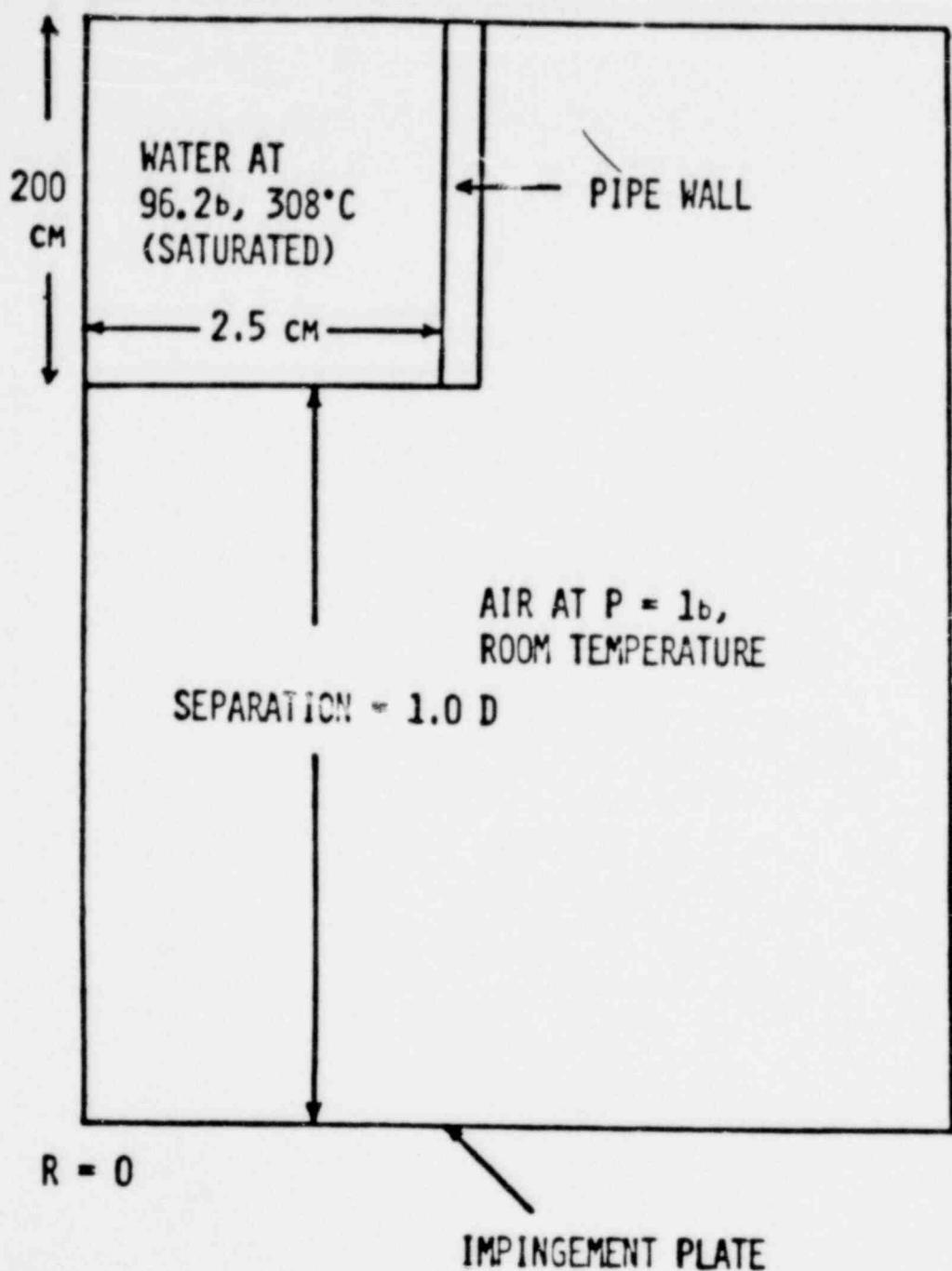
1602 529

TABLE I

<u>FACILITY</u>	<u>PRESSURE (b)</u>	<u>TEMPERATURE (°C)</u>	<u>DIAMETER CM</u>	<u>MASS FLOW KG/SEC</u>	<u>MASS FLUX KG/SEC/M²</u>	<u>COMMENTS</u>
ZION	157.23	276.7	69.85	4606	12019	PWR COLD LEG
KWU	30-100	234-311	1,2.5, 5.6.5	80	24108-1.E6	TAPERED NOZZLE, ADIABATIC EXIT PIPE
BATTELLE- FRANKFURT	140 _{MAX}	300 _{MAX}	10. (DOUBLE ENDED BREAK)	400	50929	PRESSURE VESSEL + STEAM GENERATOR. 1/64 SCALE BIBLIS, A PWR
HDR SERIES 1	88 _{MAX}	220 _{MAX}	35.0	1222 _{MAX} (STEAM)	12701	400% FLOW (ALL HEADERS ATTACHED)
SERIES 3	110 _{MAX}	310 _{MAX}	35 OR 45	1222 _{MAX} (STEAM)	7683 OR 12701	400% FLOW

1602 330

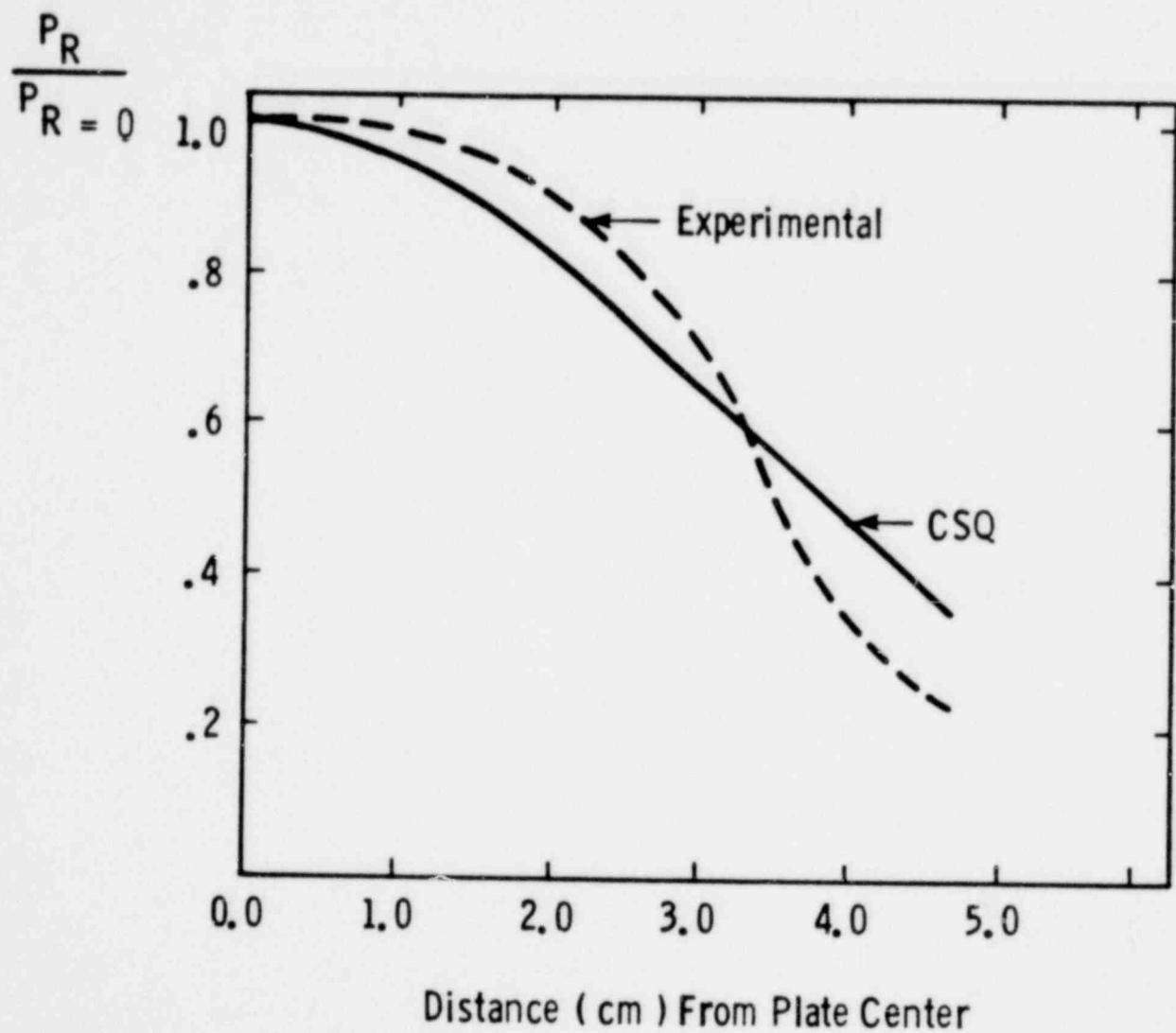
CSQ PIPE BLOWDOWN MODEL
OF KWU TEST 6



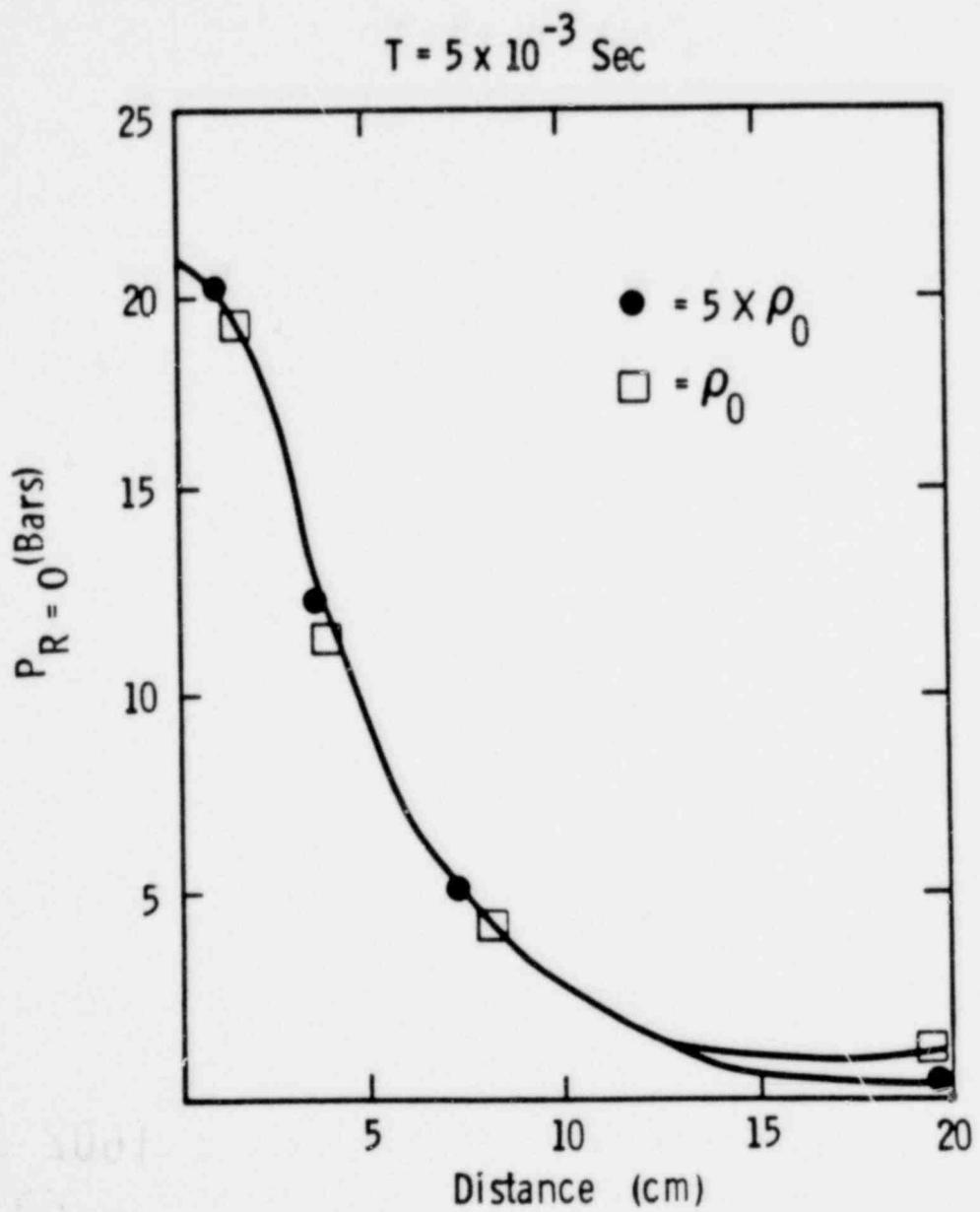
1602 331

Radial Pressure At $T = .003$ sec.

Separation = 1.0 D

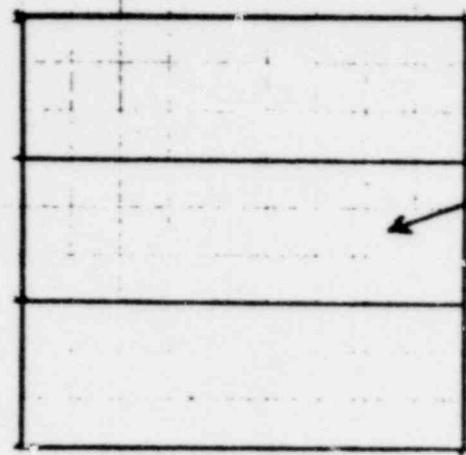


1602 332

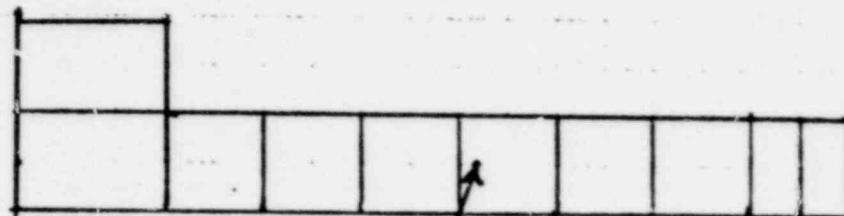


EFFECT OF AIR DENSITY - CSQ

1602 333



ACCUMULATOR



PIPE



BREAK

TRAC-PIA PIPE MODEL

1602 334

TRAC-PIA AND KWU TEST COMPARISONS

TEST	PRESSURE (B)	BREAK FLOW (KG/SEC)			STEADY-STATE IMPINGEMENT LOAD (NT)				MOODY MODEL	SEPARATED FLOW
		EXP	TRAC	MOODY	EXP MEASURED RECOIL**	EXP ORIFICE	TRAC	HOMOGENEOUS FLOW		
NW 25										
3	99.4	15.7	16.5	19.2	2520***	5288	5080	6900	7300	
5	52.4	12.2	10.2	18.0	2765***	3145	2590	3000	3200	
NW 50										
6	96.2	56.4	63.6	76.6	15428	19740	19070 17400	23600 20000****	24900	
NW 65										
3	98.7	75.4	106.9	129.6	26041	28233	32440	41000	42500	
4	53.1	53.7	70.8	121.5	14518	16102	16600	19000	22000	

*MODEL WITH FRICTION

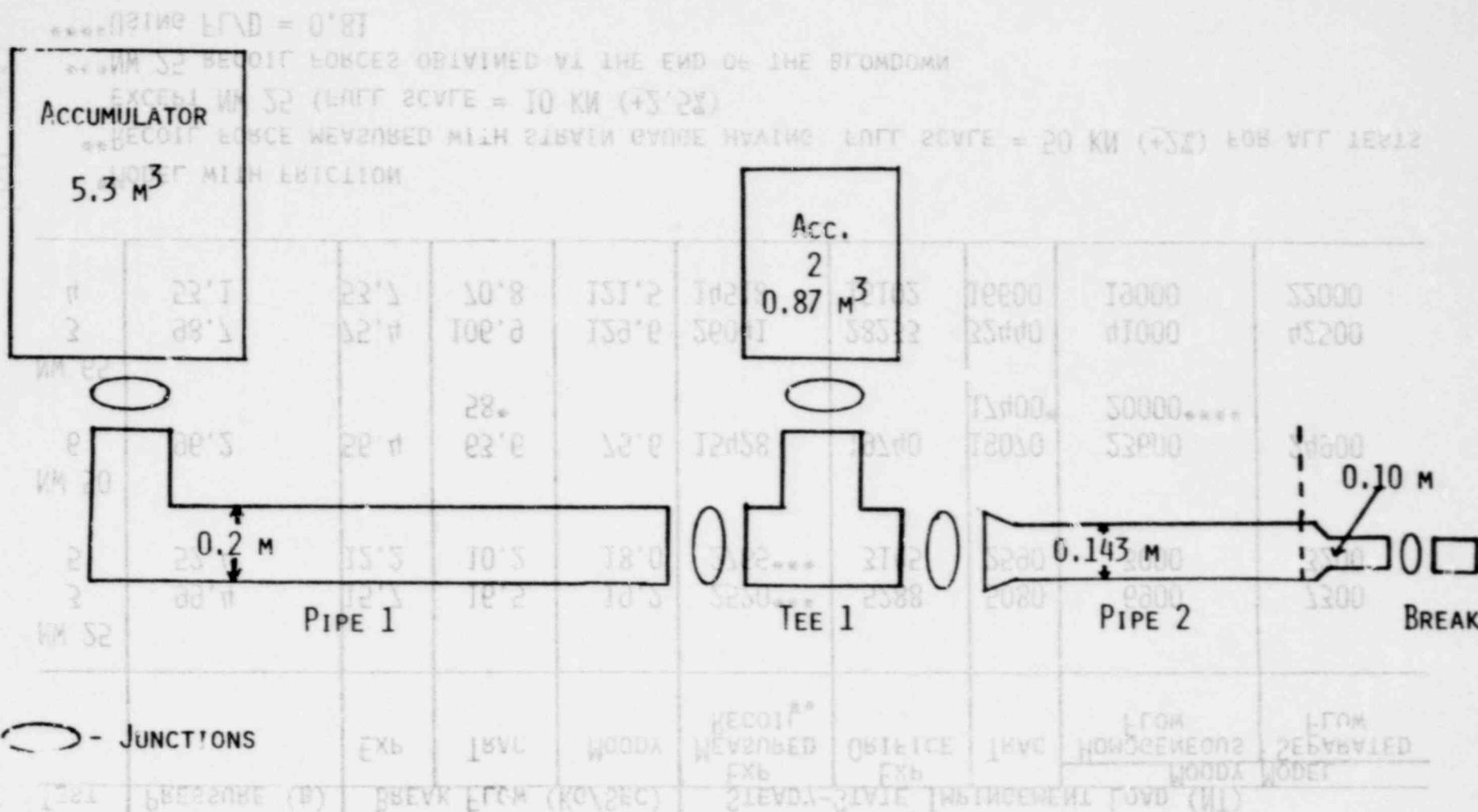
**RECOIL FORCE MEASURED WITH STRAIN GAUGE HAVING FULL SCALE = 50 KN (+2%) FOR ALL TESTS
EXCEPT NW 25 (FULL SCALE = 10 KN (+2.5%))

***NW 25 RECOIL FORCES OBTAINED AT THE END OF THE BLOWDOWN

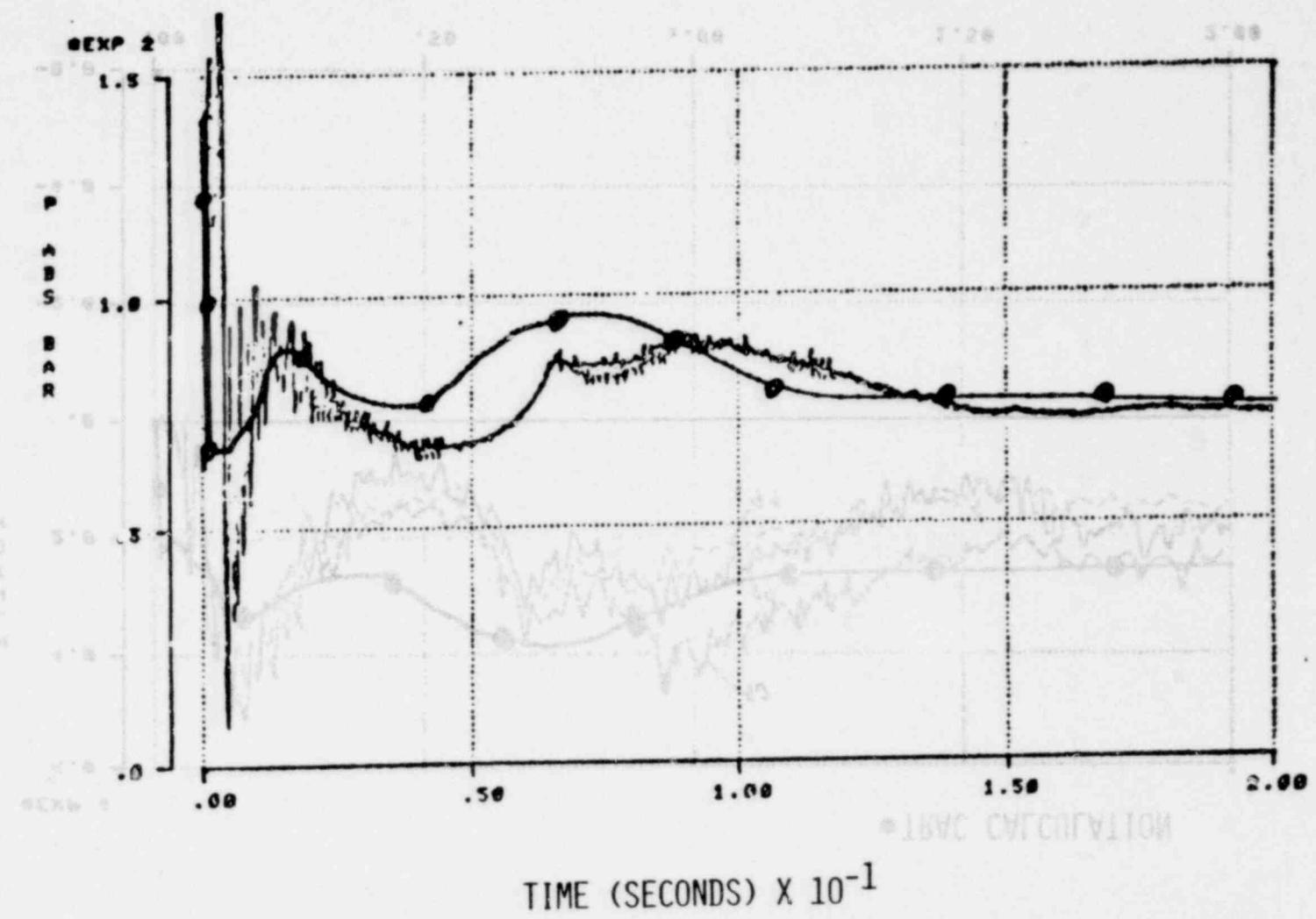
****USING FL/D = 0.81

1602
335

TRAC 6 COMPONENT RS-50 MODEL

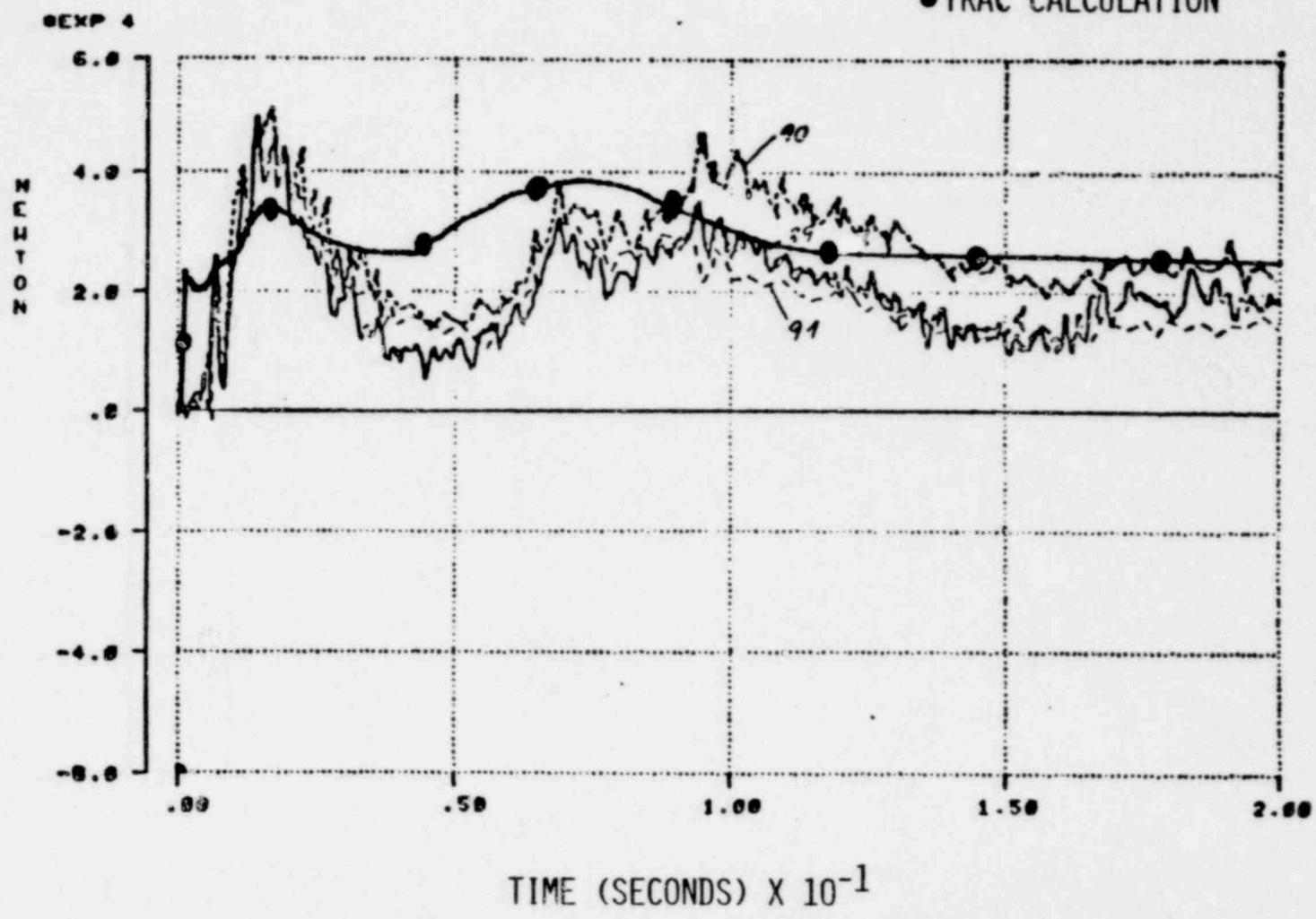


• TRAC CALCULATION



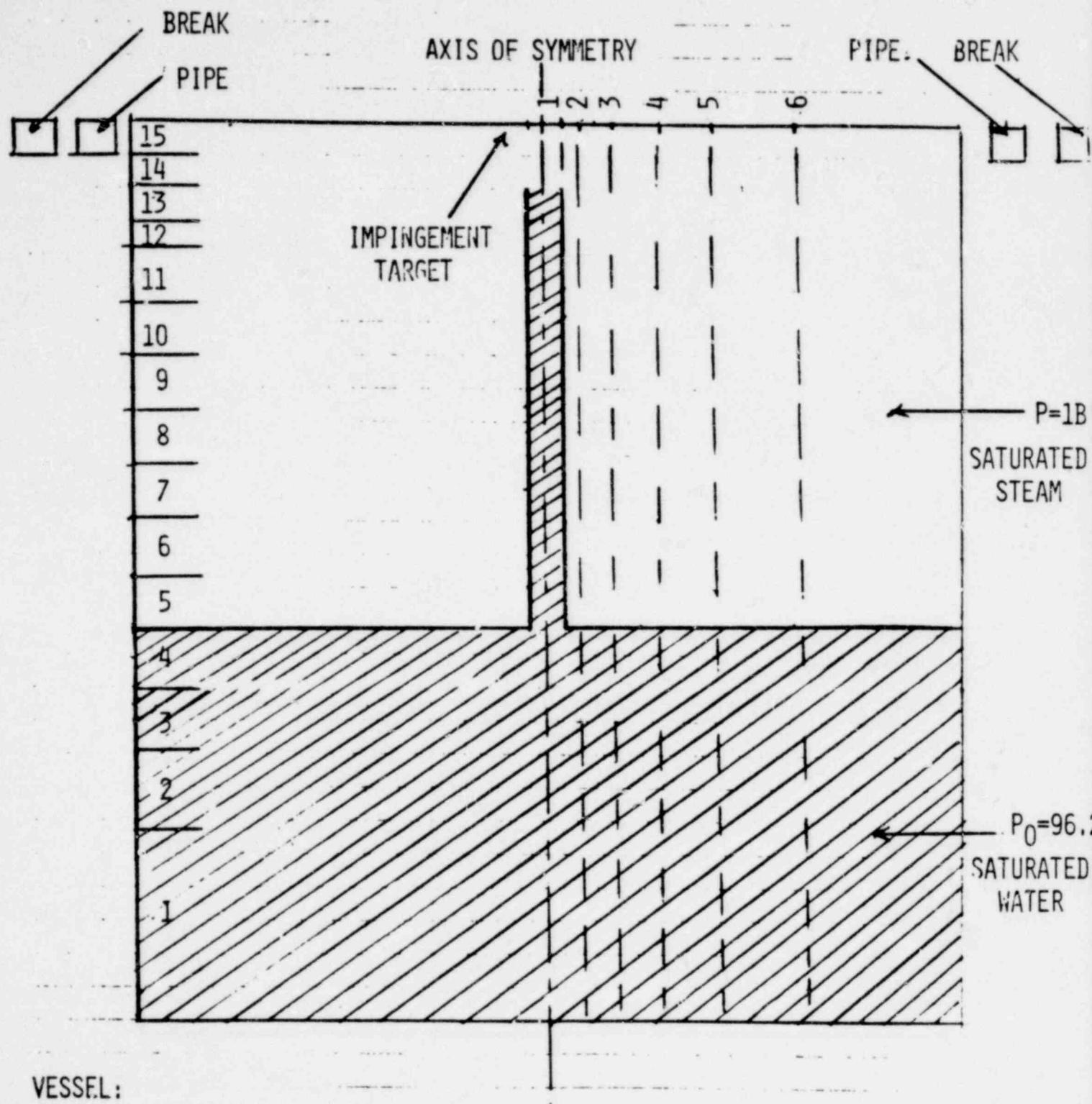
1602 337

• TRAC CALCULATION



1602 338

— TRAC-P1A KWU MODEL



VESSEL:

15 AXIAL SECTIONS

7 RINGS

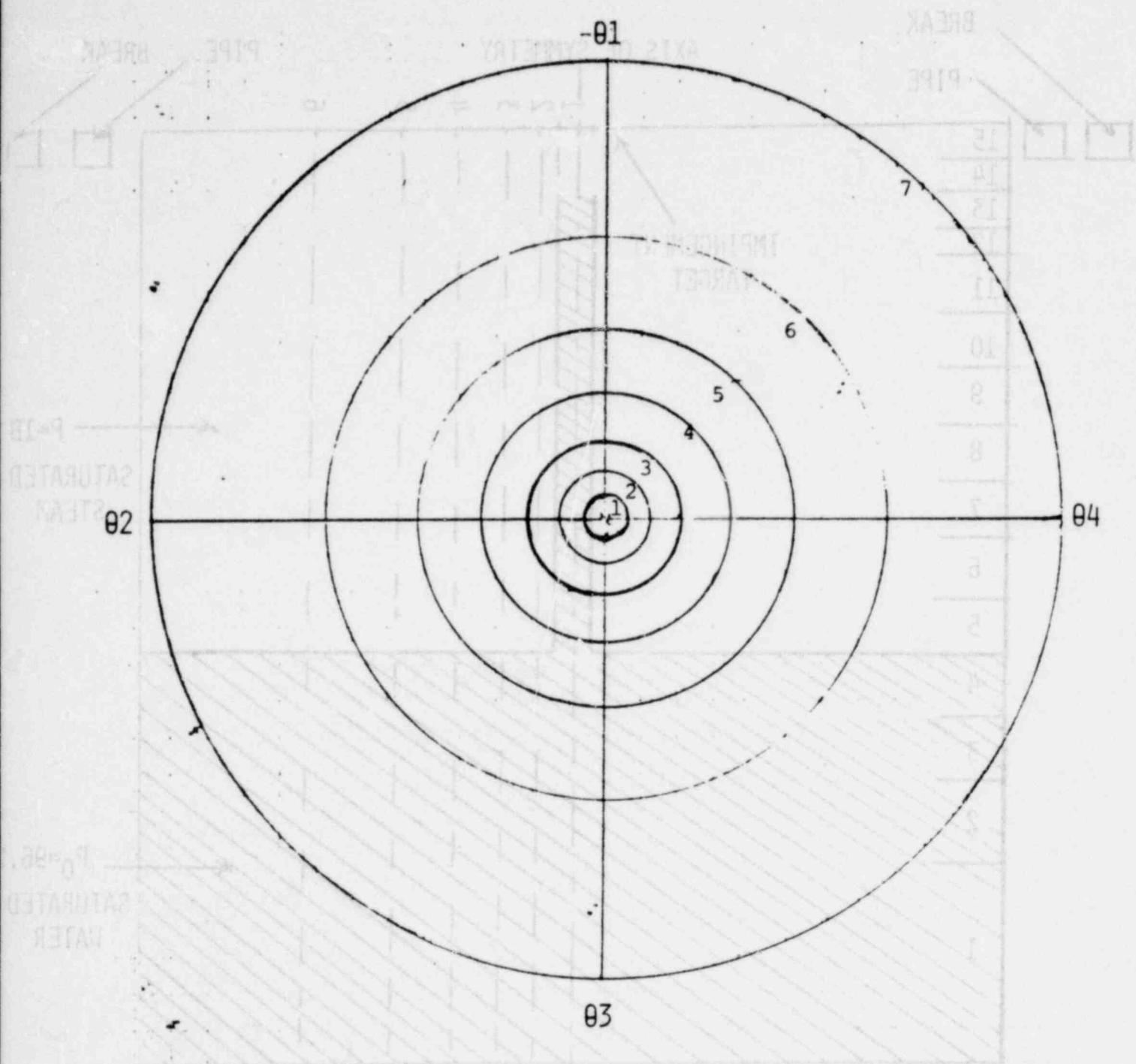
4 ANGLES

4 PIPES

4 BREAKS

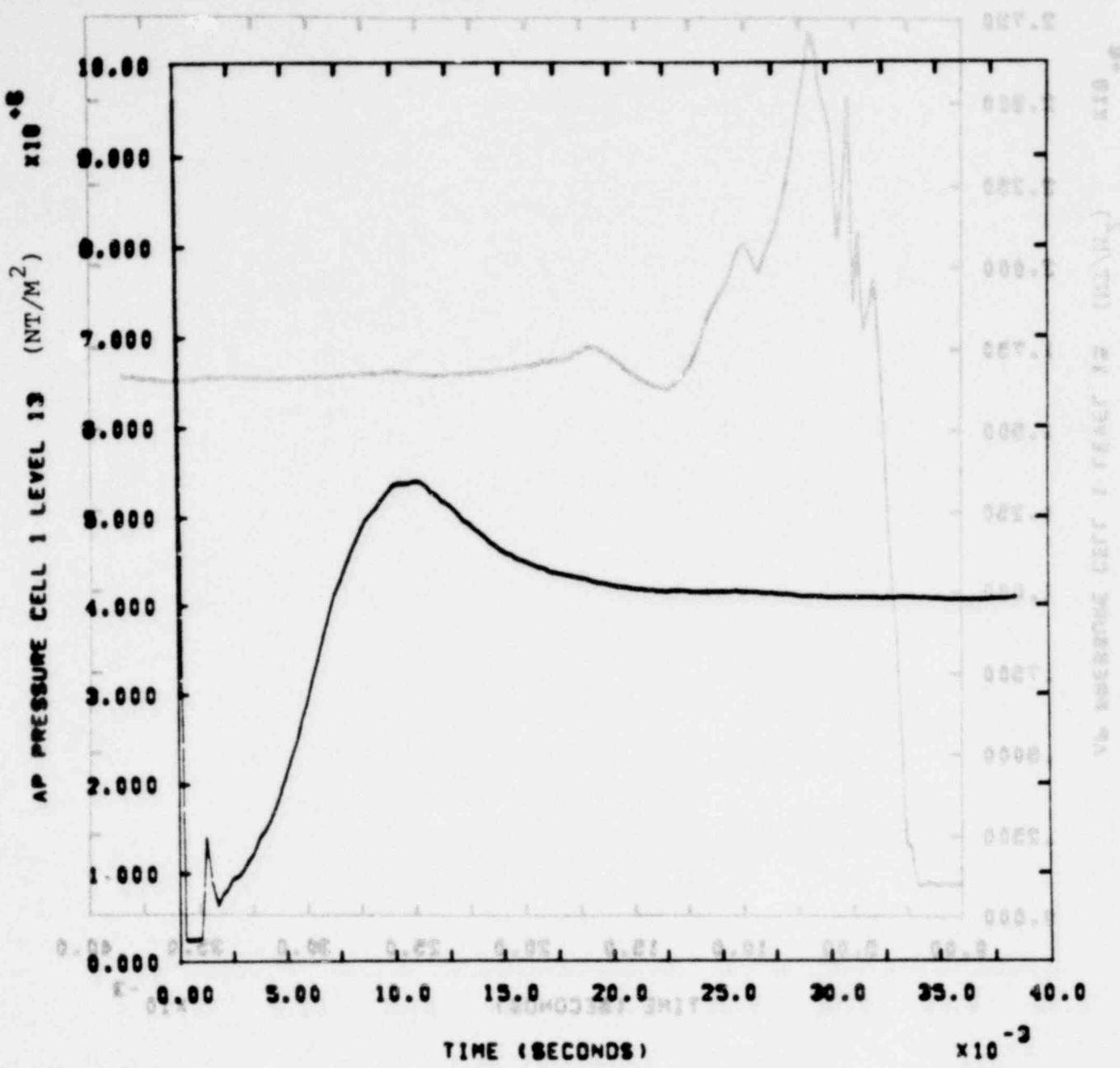
1602 339

TOP VIEW OF VESSEL SHOWING
4 ANGLES, 7 RINGS MODEL

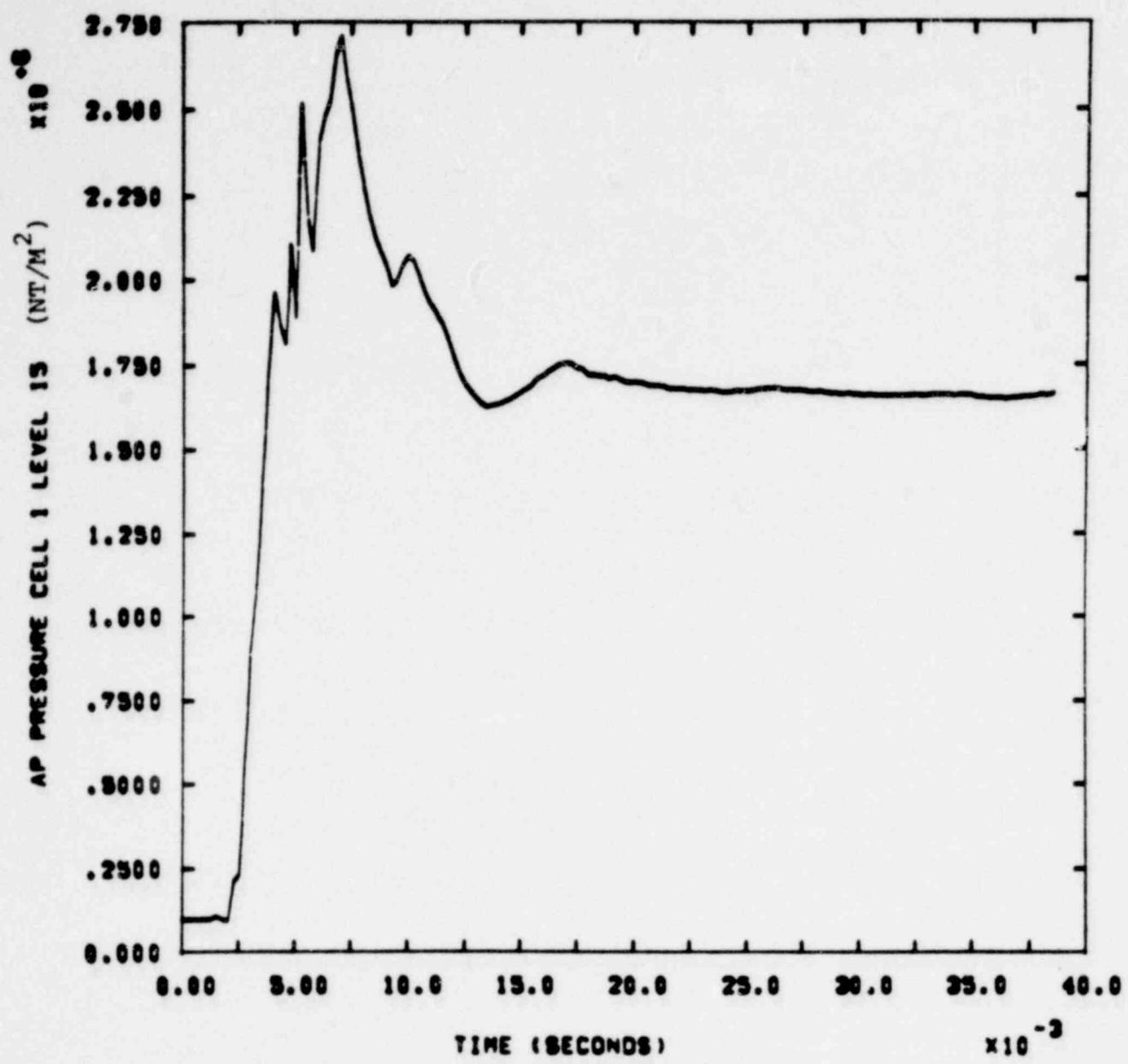


TOP VIEW OF VESSEL SHOWING
4 ANGLES, 7 RINGS MODEL

1602 340



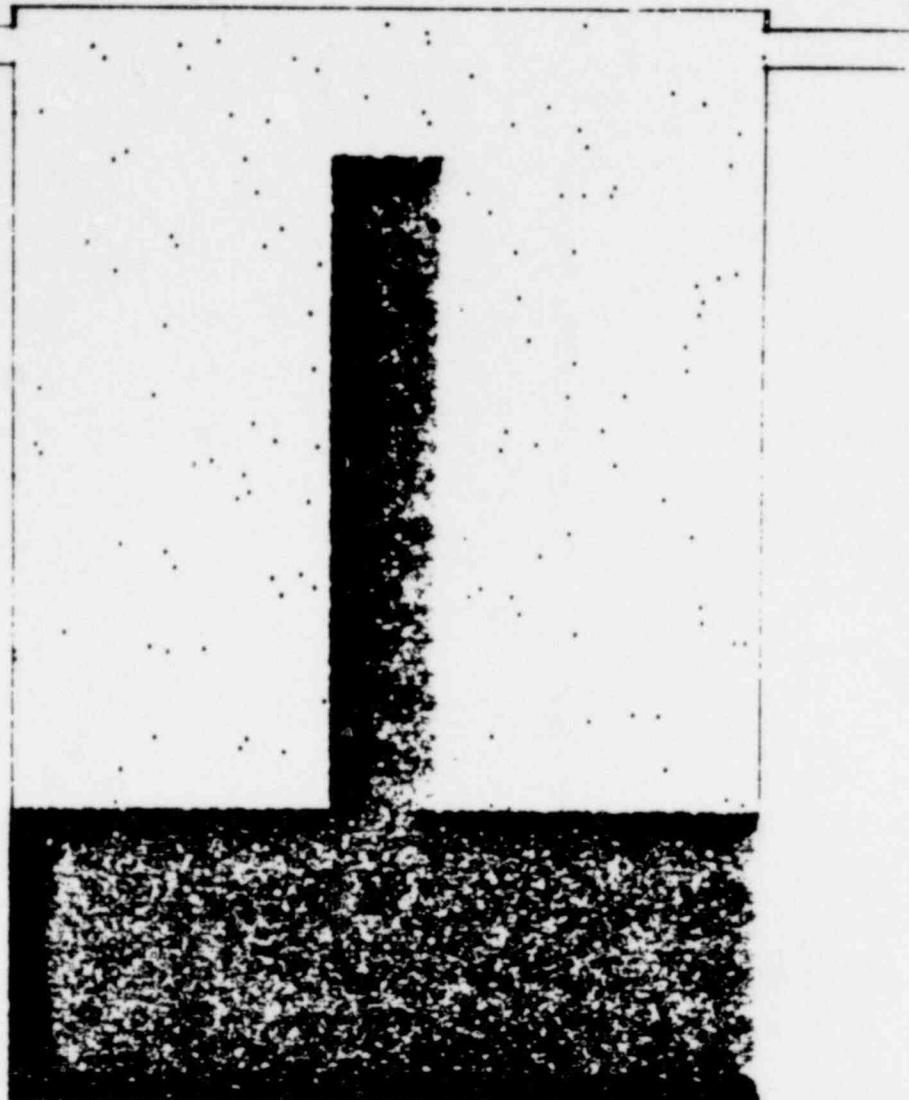
1602 341



1602 342

148 8081

RH - MIXTURE DENSITY .0.0 TO 720.0



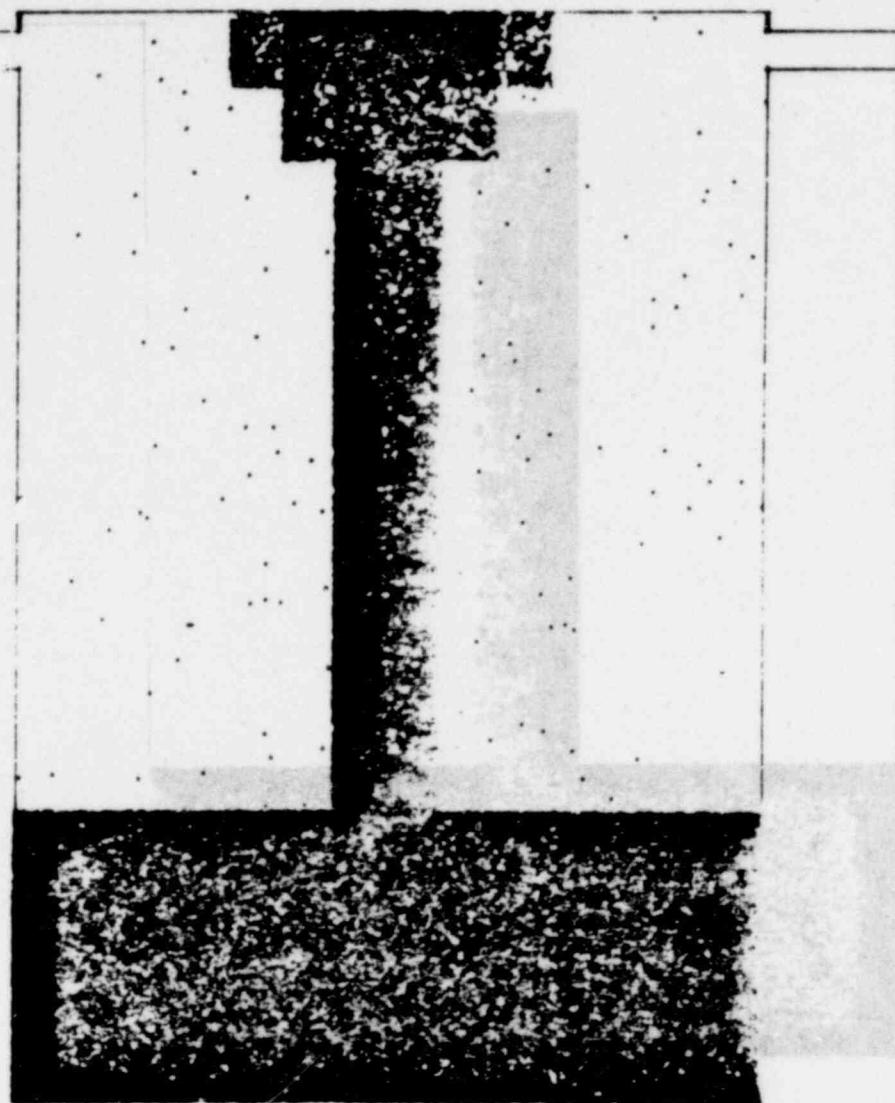
TIME = .000000 SECONDS

TRAC-P1A DENSITY

POOR ORIGINAL

1602 343

RH - MIXTURE DENSITY. 0.0 TO 720.0



TIME = .005000 SECONDS

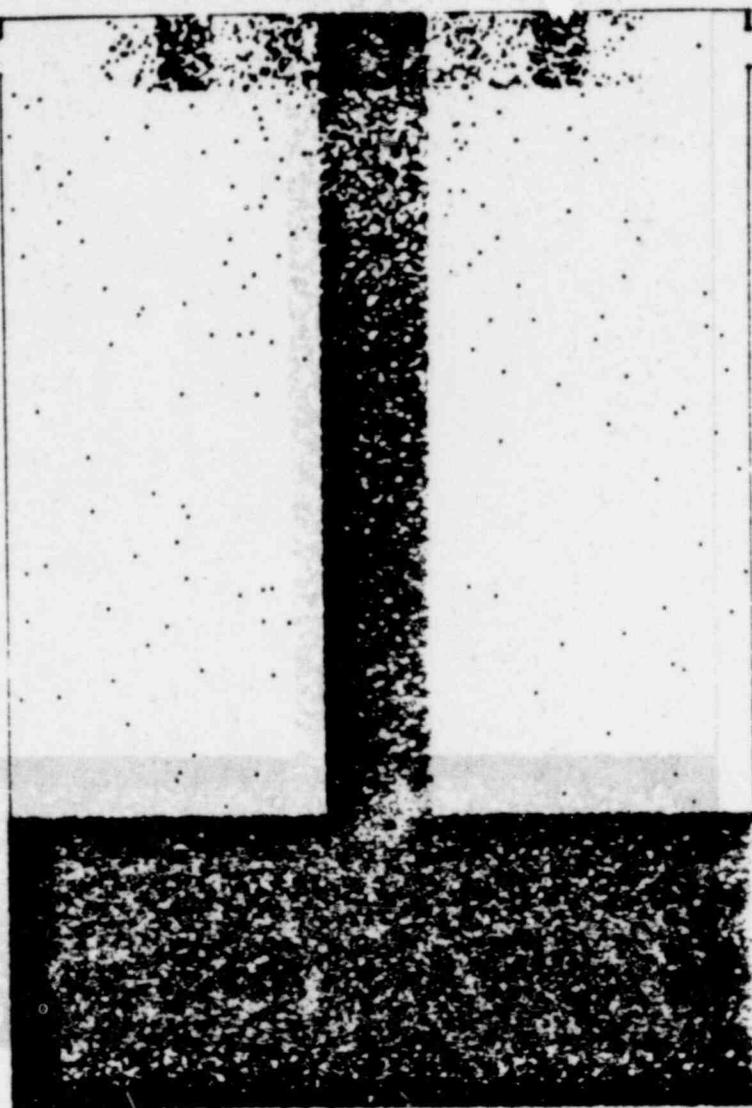
TRAC-P1A DENSITY

1602 344

TRAC-P1A DENSITY

POOR ORIGINAL

RH - MIXTURE DENSITY. 0.0 TO 720.0



TIME = .010023 SECONDS

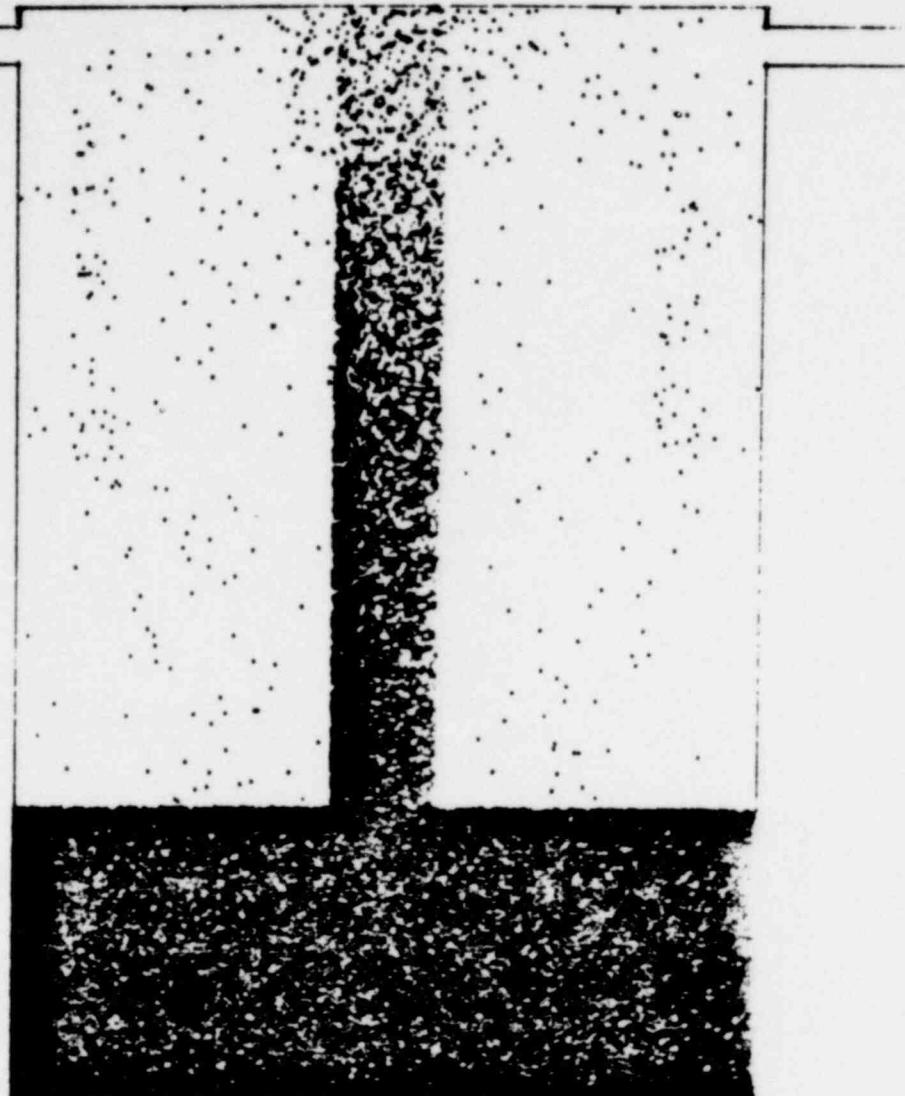
TRAC-PIA DENSITY

TRAC-PIA DENSITY

1602 345

POOR ORIGINAL

RH - MIXTURE DENSITY 0.0 TO 7200

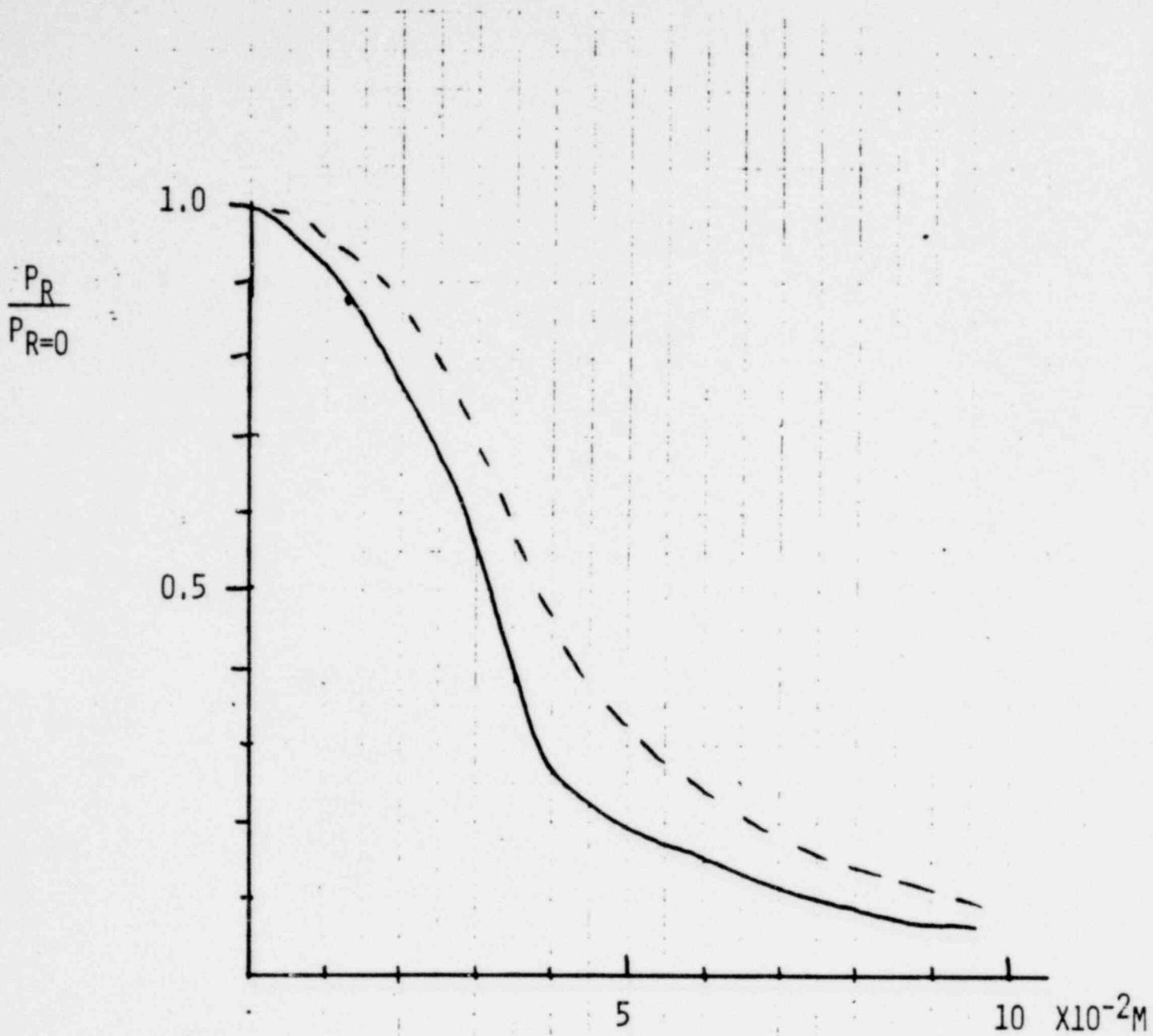


TIME = .035001 SECONDS

POOR ORIGINAL

TRAC-P1A DENSITY

1602 346



TRAC-PIA PLATE STAGNATION PRESSURE

1602 347