Exhibit C NEP-12-02 Revision 0 page 1 of 2

COMMONWEALTH EDISON COMPANY CALCULATION TITLE PAGE

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	CALCULATION NO. QDC-1400-M-0080 PROJECT NO.: N/A PAGE NO.: 1
	SAFETY RELATED REGULATORY RELATED NON-SAFETY RELATED
	<u>CALCULATION TITLE:</u> Pressure Locking Calculation for Core Spray System Valve MOV 1-1402-25A
	STATION/UNIT: Quad Cities/1 SYSTEM ABBREVIATION: LPCS
	EQUIPMENT NO.: (IF APPL.) MOV 1-1402-25A
	REV: 0 STATUS: QA SERIAL NO. OR CHRON NO. N/A DATE:
	PREPARED BY: <u>K. Higgins</u> REVISION SUMMARY: Original Issue DO ANY ASSUMPTIONS IN THIS CALCULATION REQUIRE LATER VERIFICATION YES = NO KEE ASSUMP. 4
	REVIEWED BY: <u>J. Kelly</u> Jel 7 Pell 11-9-95
	REVIEW METHOD: Detailed review COMMENTS (C OR NC): <u>// _</u>
	APPROVED BY: LA Content WOParren 11/9/95
	9602200285 960213 PDR ADDCK 05000237 PDR PDR

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COMMONWEALTH EDISON COMPANY CALCULATION REVISION PAGE

CALCULATION NO.	QDC-1400-M-0080	PAGE NO.: 2
REV: STATUS:	QA SERIAL NO. OR CHRON NO.	DATE:
PREPARED BY: REVISION SUMMARY:		DATE:
REVIEWED BY: REVIEW METHOD:		DATE: COMMENTS (C OR NC):
REV: STATUS:	QA SERIAL NO. OR CHRON NO.	DATE:
PREPARED BY: REVISION SUMMARY:		DATE:
REVIEWED BY: REVIEW METHOD:		DATE: COMMENTS (C OR NC):
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Exhibit D NEP-12-02 Revision 0

COMMONWEALTH EDISON COMPANY

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CALCULATION NO.	QDC-1400-M-0080	PROJECT NO.

N/A

I. PURPOSE/OBJECTIVE

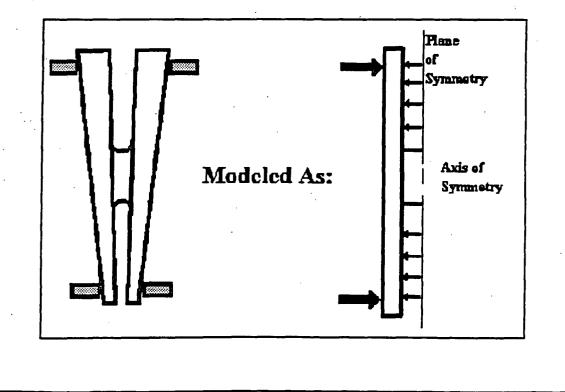
Valve MOV 1-1402-25A, which is installed in the Core Spray System at Quad Cities, has been determined to be susceptible to the pressure locking phenomena. The purpose of this calculation is to determine that the valve will open when called upon by design during a pressure locking event by calculating the force necessary to overcome pressure locking and the available Motor/Gearing Capability (MGC) of the MOV.

II. METHODOLOGY AND ACCEPTANCE CRITERIA

The methodology for calculating the thrust required to open the MOVs under the pressure locking scenario is based on the Reference 1 (Roark's) engineering handbook. This methodology has been previously applied by ComEd in the References 2 and 10 calculations. The methodology determines the total force required to open the valve under a pressure locking scenario by solving for the four components to this required force. The four components of the force are the Pressure Locking Component, the static unseating component, the piston effect component, and the "reverse piston effect" component. These components are determined using the following steps.

Pressure Locking Component of Force Required to Open the Valve

The valve disk is modeled as two plates attached at the center by a hub which is concentric with the valve disk. A plane of symmetry is assumed between the valve disks. This plane of symmetry is considered fixed in the analysis.

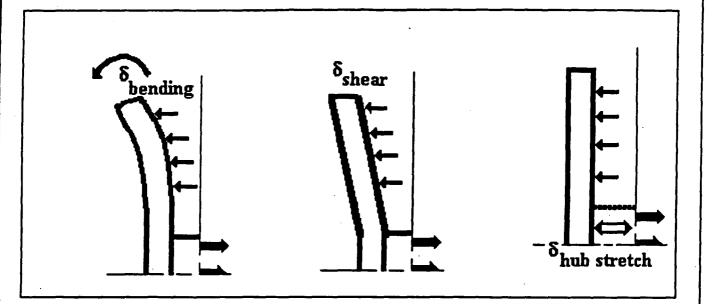


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The pressure force is assumed to act uniformly upon the inner surface of the disk between the hub diameter and the outer disk diameter. The outer edge of the disk is assumed to be unimpeded and allowed to deflect away from the pressure force. In addition, the disk hub is allowed to stretch. The total displacement at the outer edge of the valve disk due to shear and bending and due to hub stretch are calculated using the reference 1 equations.



An evenly distributed force is assumed to act between the valve seat and the outer edge of the valve disk. This force acts to deflect the outer diameter of the valve disk inward and to compress the disk hub. The pressure force is reacted to by an increase in this contact force between the valve disk and seats. The valve body seats are conservatively assumed to be fixed. Therefore, the deflection due to the known pressure load must be balanced by the deflection due to the unknown seat load. The deflection due to the pressure force is first calculated. Then, the reference 1 equations are used to determine the contact force between the seat and disk which results in a deflection which is equal and opposite to the deflection due to the pressure force. Because the disk varies in thickness, the average thickness is used for purposes of the calculation.

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The coefficient of friction between the seat and disk is determined based on the open valve factor from a DP test. The stem force required to overcome the contact load between the seat and disk which opposes the pressure force is equal to:

(seat load) x [(seat mu) cos(seat angle) - sin(seat angle)] x 2 (for two disk faces).

Static Unseating Force

The static unseating force represent the open packing load and pullout force due to wedging of the valve disk during closure. These loads are superimposed on the loads due to the pressure forces which occur during pressure locking. The value for this load is based on static test data for the MOVs.

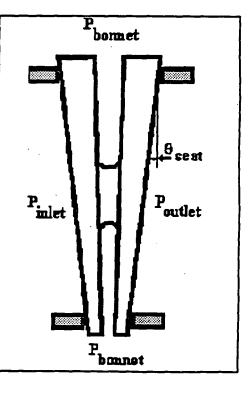
Piston Effect

The piston effect due to valve internal pressure exceeding outside pressure is calculated using the standard industry equation. This force assists movement of the valve stem in the open direction.

$F_{pirtumetres} = \frac{.s}{4} \times D_{run}^2 \times (P_{human} - P_{out})$

"Reverse Piston Effect" (F_{vert})

The reverse piston effect is the term used in this calculation to refer to the pressure force acting downward against the valve disk. This force is equal to the differential pressure across the valve disk times the area of the valve disk times the sine of the seat angle times 2 (for two disk faces).



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Total Force Required to Overcome Pressure Locking

As mentioned previously, the total stem force (tension) required to overcome pressure locking is the sum of the four components discussed above. All of the terms are positive with the exception of the piston effect component.

Determination of Motor Gearing Capability

Next the motor gearing capability available to overcome static unseating forces is determined using the statically measured stem factor, the pullout efficiency, the temperature factor, and the ComEd motor test data (for breakdown torque and voltage factor), Reference 5.

Determination of Open Valve Factor

The open valve factor is calculated by based on the open DP load. This load is determined by using the equation below: The O10 thrust is measured in the region of the trace during which the valve disk is sliding on the valve seat (prior to flow initiation). This thrust is based on the O4 zero since the valve is effectively closed at O10. The open running thrust is measured at the end of the open stroke and is referenced to the C3 zero since the valve is nearly fully open at the point at which the open running load is measured. The Line Pressure adjustment term in the equation accounts for the fact that the piston effect decreases during the opening valve stroke.

$$VF_{max} = \frac{O10_{shart} - Ruming_{shart} + \frac{\pi}{4}D_{rom}^{2}(O10_{shartarus} - Ruming_{hartarus})}{DP \times \frac{\pi}{4}D_{rom}^{2}}$$

Enhanced Capability Evaluation

The enhanced capability evaluation uses the measured overall MOV efficiency to predict the required motor torque during the pressure lock condition. This overall MOV efficiency is determined by dividing the actual motor torque during the latest static test by the measured pullout thrust. This ratio (MOV efficiency) is then multiplied by the estimated pressure lock pullout force to determine the required motor torque during pressure lock pullout. The available motor torque is set equal to the motor breakdown torque from the ComEd motor test data, Reference 5.

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Acceptance Criteria

The margin between one time structural limit and required thrust is calculated. A margin of 15% or greater is required. This margin accounts for equipment inaccuracy and degradation.

The margin between MGC and required thrust is calculated. A margin of 20% or greater is required. This margin accounts for equipment inaccuracy and degradation.

If the enhanced capability evaluation is applied, a margin of 40% or greater is required. This margin accounts for voltage, current, and thrust measurement inaccuracies from the static test and a possible reduction in overall MOV efficiency from the static condition to the estimated pullout condition.

III. ASSUMPTIONS

- 1. The valve disk is assumed to act as two ideal disks connected by a hub. The equations in reference 1 are assumed to conservatively model the actual load due to pressure forces. This assumption is considered conservative since inspection of the disk drawings show large fillets between the disk hub and seats which should make the valve disk stiffer than assumed in the reference 1 equations.
- 2. The coefficient of friction between the valve seat and disk is assumed to be the same under pressure locking conditions as it is under DP conditions. This assumption (in combination with assumption 1) is considered to be justified based on bench marking of the calculation against ComEd and EPRI pressure locking test data for similar flex-wedge gate valves.
- 3. The upstream, downstream, and bonnet pressure values are based on a scenario in which the valve bonnet is pressurized to reactor pressure by leakage past adjacent check valves. A LOCA occurs which causes the reactor pressure to drop off to 325 psig. The LPCS pump comes up to speed and the subject valve receives a signal to open simultaneously. Total dynamic head for the CS pump is approximately 380 psig. The pressure values are based on a review of the UFSAR for Quad Cities and the CS Pump Manual, C0048. See Reference 6.
- 4. A valve temperature of 185 *F is assumed for the purpose of this calculation. This temperature is taken from reference 12 and is based on a one time limitation of the disk ears. The valve temperature typically does not exceed 150 *F. The assumed temperature will require verification.

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CAL	CULATIO	ON NO.	QDC-1400			PROJECT 1	NO. <i>N/A</i>	PAGE NO.9	
IV.		GN INPU						- 	
	1.	Valve Disk Geometry information is based on the Reference 4, Faxes from the Crane-Aloyco Valve Company. (Attachment A)							
	2. Motor Data is taken from the Reference 5 report and RSMDS, Reference 7.								
	3.	Static a	nd DP diag	nostic test da	ata is taker	from the m	iost recent di	agnostic tests.	
V.	REFE	RENCES	5						
	1.	Sixth E	dition of Rc	xark's Formu	ulas for Sti	ress and Stra	in.		
	2. MPR Calculations 101-013-1, "Effect of Bonnet Pressure on Disc to Seat Contact Load", dated 3/23/95; and 101-013-4, "Estimate of Valve Unseating Force as Function of Bonnet Pressure", dated 3/23/95.								
	3. NMAC Report NP-6660-D, " Application Guide For Motor Operated Valves" .							ted Valves".	
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CALCULAT	ION NO. QDC-1400-M-0080	PROJECT NO. N/A	PAGE NO.10							
4.	4. Crane Telecopies from Dave Dwyer and Bruce Harry to Brian Bunte (ComEd) dated 5/3/95 and 6/16/95, Attachment A.									
5.	5. ComEd White Paper 125, MOV-WP-125, Rev. 2, 10/4/95									
6.	6. UFSAR Section 6.3.2.1.2, Fig 6.3-3, Tbl 4.1-3 and Core Spray Pump Manual C0048									
7.	ComEd Rising Stem MOV Data Sheet,	, 1-1402-25A, 03/03/95, 11	:05							
8.	Thrust values are taken from static VO VOTES Test 38 performed 08/12/94	TES Test 37 performed 08/	07/94 and DP							
9.	EMS Calculation CE-DR-030, "Pressur Operated Valves", dated 6/13/95.	re Locking Analysis of Dres	sden Motor							
10.	ComEd Calculation NED-M-MSD-182 Quad Cities Injection Valves Susceptible		- 1							
11.	Crane-Aloyco Valve Drawing CA0079). •								
12.	EMS Letter, EMSP-95-324, 9-15-94									
VI. CALCUL	ATIONS									
The f	following is provided for MOV 1-1402-2	5A.								
Math	Cad calculation of:									
1) 2) 3)	2) the available motor gearing capability to unseat while pressure locked									
	· · · ·									
	,									

CALCULATION NO. QDC-14	<i>w-m-</i> 0080	PROJECT NO. N	/A PAGE NO. 11
VI QCNPS Valve 1 INPUTS:	-1402-25A		
Bonnet Pressure Upstream Pressure Downstream Pressure	P _{bonnet} := 1005 psi P _{up} := 380 psi P _{down} := 325 psi	Reference 6 Ref. 6 & Assum. 3 Ref. 6 & Assum. 3	
Disk Thickness, Avg Seat Radius Hub Radius Hub Length Seat Angle Poisson's Ratio (disk) Mod. of Elast. (disk)	t := 1.69 in a := 4.36 in b := 1.25 in L := 1.625 in theta := 5 deg v := 0.3 E := 27.6 $\cdot 10^{6}$ psi	Reference 4 Reference 7 Reference 4 Reference 4 Reference 7 Stain. Steel, Ref. 1 & 11 Stain. Steel, Ref. 1 & 11	
Static Pullout Force (Test 37) O10 Thrust (DP test 38) Open Run Thrust (DP) DP LP (valve closed) LP (valve open) Stem Diameter	F po := 39071.lbf O10 := 13082.lbf Run := 504.lbf DPtest := 350.psi LPclose := 350.psi LPopen := 0.psi D stem := 1.875.in	Reference 8 Reference 8, Att. B Reference 7	•

VALVE FACTOR CALCULATION

Valve Factor:

$$(O10 - Run) + \frac{\pi}{4} D_{stem}^{2} (LPclose - LPopen)$$

VF := _____

 π (a)² DPtest

VF = 0.648

CCoefficient of friction between disk and seat: (I(Reference 3)

 $mu := VF \cdot \frac{\cos(\text{theta})}{1 - VF \cdot \sin(\text{theta})}$

mu = 0.684

PRESSURE FORCE CALCULATIONS

Average DP across disks:

$$DPavg := P_{bonnet} - \frac{P_{up} + P_{down}}{2}$$

DPavg = 652.5 •psi

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Disk Stiffness Constants (Reference 1, Table 24)		
$D := \frac{E \cdot (t)^{3}}{12 \cdot (1 - v^{2})}$	$D = 1.22 \cdot 10^7$ ·lbf in	
$\mathbf{G} := \frac{\mathbf{E}}{2 \cdot (1 + \mathbf{v})}$	$G = 1.062 \cdot 10^7 \text{ psi}$	
Geometry Factors: (Reference 1, Table 24)		
$\mathbf{C}_{2} := \frac{1}{4} \cdot \left[1 - \left(\frac{\mathbf{b}}{\mathbf{a}} \right)^{2} \cdot \left(1 + 2 \cdot \ln \left(\frac{\mathbf{a}}{\mathbf{b}} \right) \right) \right]$	$C_2 = 0.178$	
$C_{3} := \frac{b}{4 \cdot a} \cdot \left[\left[\left(\frac{b}{a} \right)^{2} + 1 \right] \cdot \ln \left(\frac{a}{b} \right) + \left(\frac{b}{a} \right)^{2} - 1 \right]$	$C_3 = 0.031$	
$C_{8} = \frac{1}{2} \cdot \left[1 + v + (1 - v) \cdot \left(\frac{b}{a} \right)^{2} \right]$	C ₈ = 0.679	
$C_{9} := \frac{b}{a} \cdot \left[\frac{1+v}{2} \cdot \ln\left(\frac{a}{b}\right) + \frac{1-v}{4} \cdot \left[1 - \left(\frac{b}{a}\right)^{2} \right] \right]$	C ₉ = 0.279	•
$L_{3} := \frac{a}{4 \cdot a} \cdot \left[\left[\left(\frac{a}{a} \right)^{2} + 1 \right] \cdot \ln \left(\frac{a}{a} \right) + \left(\frac{a}{a} \right)^{2} - 1 \right]$	$L_{3} = 0$	
$L_{9} := \frac{a}{a} \cdot \left[\frac{1+v}{2} \cdot \ln\left(\frac{a}{a}\right) + \frac{1-v}{4} \cdot \left[1 - \left(\frac{a}{a}\right)^{2} \right] \right]$	$L_{9} = 0$	
$L_{11} := \frac{1}{64} \cdot \left[1 + 4 \cdot \left(\frac{b}{a}\right)^2 - 5 \cdot \left(\frac{b}{a}\right)^4 - 4 \cdot \left(\frac{b}{a}\right)^2 \cdot \left[2 + \left(\frac{b}{a}\right)^2 \right] \right]$	$\left[\ln\left(\frac{\mathbf{a}}{\mathbf{b}}\right) \right] \qquad \qquad \mathbf{L}_{11} = 0.007$	•
$L_{17} = \frac{1}{4} \cdot \left[1 - \frac{1 - v}{4} \cdot \left[1 - \left(\frac{b}{a} \right)^4 \right] - \left(\frac{b}{a} \right)^2 \cdot \left[1 + (1 + v) \cdot \frac{b}{a} \right] \right]$	$\ln\left(\frac{a}{b}\right) \end{bmatrix} \qquad \qquad L_{17} = 0.153$;
Moment (Reference 1, Table 24, Case 2L)		
$M_{rb} := \frac{-DPavg \cdot a^2}{C_8} \cdot \left[\frac{C_9}{2 \cdot a \cdot b} \cdot (a^2 - b^2) - L_{17} \right]$	$M_{rb} = -5.368 \cdot 10^3 \cdot lbf$	
$Q_{b} := \frac{DPavg}{2 \cdot b} \cdot (a^{2} - b^{2})$	$Q_{b} = 4.554 \cdot 10^{3} \cdot \frac{lbf}{in}$	
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Deflection due to pressure and bending: (Refere	ence 1, Table 24, C	ase 2L)	
$y_{bq} = M_{rb} \frac{a^2}{D} C_2 + Q_b \frac{a^3}{D} C_3 - \frac{DPavg a^4}{D} L_1$	1	y _{bq} = -6.597	•10 ⁴ •in
Deflection due to pressure and shear stress: (R	leference 1, Table	25, Case 2L)	
$K_{sa} := -0.3 \cdot \left[2 \cdot \ln \left(\frac{a}{b} \right) - 1 + \left(\frac{b}{a} \right)^2 \right]$		K _{sa} = -0.474	
$y_{sq} = \frac{K_{sa} \cdot DPavg \cdot a^2}{t \cdot G}$		y _{sq} = -3.279•	10 ⁻⁴ •in
Deflection due to hub stretch (from center of hub to	o disk):		
$P_{\text{force}} = 3.1416 \cdot (a^2 - b^2) \cdot DPavg$		$P_{force} = 3.5$	76•10 ⁴ •lbf
$y_{\text{stretch}} = \frac{P_{\text{force}}}{3.1416 \text{ b}^2 (2 \cdot \text{E})}$	•	$y_{\text{stretch}} = 2.$	145•10 ⁻⁴ •in
Total Deflection due to pressure forces:			
$y_q := y_{bq} + y_{sq} - y_{stretch}$		$y_{q} = -0.001$	•in
Deflection due to seat contact force and shear stre		eference 1, Tal ase 1L)	ble 25,
y _{sw} := - $\left[\frac{1.2 \cdot \left(\frac{a}{a}\right) \cdot \ln\left(\frac{a}{b}\right) \cdot a}{t \cdot G}\right]$ (per lbf/in)		y _{sw} =-3.644•	$10^{-7} \cdot \frac{\text{in}}{\left(\frac{\text{lbf}}{\text{in}}\right)}$
Deflection due to seat contact force and bending (p	er lbf/in.): (Refer	ence 1, Table 2	4 ,
y by $= -\left(\frac{a^3}{D}\right) \cdot \left[\left(\frac{C_2}{C_8}\right) \cdot \left[\left(\frac{a \cdot C_9}{b}\right) - L_9\right] - \left[\left(\frac{a}{b}\right) \cdot C_3\right]\right]$ (per lbf/in)	$\left + L_3 \right $	^y bw = -9.965•	$10^{-7} \cdot \frac{\text{in}}{\left(\frac{\text{lbf}}{1}\right)}$
Deflection due to hub compression (per lbf/in), (fror			(m)
$\frac{y_{\text{compr}}}{(\text{per lbf/in})} = \frac{2 \cdot a \cdot \pi}{3.1416 \cdot b^2} \cdot \frac{L}{(2 \cdot E)}$		y _{compr} = 1.643	$\cdot 10^{-7} \cdot \frac{\text{in}}{\left(\frac{\text{lbf}}{1}\right)}$
Total deflection due to seat contact force (per lbf/in.			\ in /
$y_w = y_{bw} + y_{sw} - y_{compr}$ (per lbf/in)	. •	y _w =-1.525•1	$0^{-6} \cdot \frac{\text{in}}{\left(\frac{\text{lbf}}{\text{in}}\right)}$

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Seat Contact Force for wh from pressure forces:	ich deflection is equal prev	riously calculated deflection	in
$F_{s} := 2 \cdot \pi \cdot a \cdot \frac{y_{q}}{y_{w}}$		$F_s = 2.159 \cdot 10^{-10}$	0 ⁴ •lbf
UNSEATING FORCES	(Reference 2)	
F _{packing} is included in m	easured static pullout Forc	e	
$F_{\text{piston}} = \frac{\pi}{4} D_{\text{stem}}^2 P_{\text{b}}$	onnet	$F_{\text{piston}} = 2.775 \cdot 10^3 \cdot 1b$	f
$F_{vert} = \pi a^2 \sin(theta) (2)$	$P \cdot P_{\text{bonnet}} - P_{\text{up}} - P_{\text{down}}$	$F_{vert} = 6.792 \cdot 10^3 \cdot lbf$	
$F_{\text{preslock}} = 2 \cdot F_{\text{s}} \cdot (\text{mu cc})$	os(theta) - sin(theta))	$F_{\text{preslock}} = 2.567 \cdot 10^4 \cdot$	lbf
$F_{total} := F_{piston} + F_{ver}$	t ^{+ F} preslock ^{+ F} po	$F_{po} = 3.907 \cdot 10^4 \cdot lbf$	
$F_{total} = 6.876 \cdot 10^4 \cdot lbf$		· .	
MOTOR / GEARING CAI	PABILITY INPUTS:	· · · ·	
Motor Torque:	MR = 73.5 ft lbf	Reference 5	•
Temperature Factor:	Tf := 0.936	Reference 7	· .
Degraded Voltage:	DV := 381 volt	Reference 7	
Under Voltage Factor:	n := 2.0875	Reference 5	
Stem Factor:	SF := 0.0182 · ft	Reference 7	
Overall Ratio:	OAR := 29.44	Reference 7	
Pullout Efficiency:	Eff _{po} := 0.45	Reference 7	
Motor Torque at Pullout (from latest static test)	MT _{po} = 18.65 ft lbf	Ref 8, motor 36, pf = 0.84 @	16 A
MOTOR / GEARING CAP	ABILITY CALCULAT	IONS:	
MGC :≈ MR·Tf·OAR·Eff _p	$o \frac{\left(\frac{DV}{460 \text{ volt}}\right)^n}{SF}$	$MGC = 3.379 \cdot 10^4 \cdot lbf$	

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OPEN STRUCT	URAL LIMIT: StructuralLimit := 83505-1	of (Reference	12, Assum	n. 4)
	StructuralLimit – F MarginStructural := F total	total MarginS	tructural = (0.214
OPEN LIMIT:	Limit := (min((StructuralL	imit MGC)))		
MARGIN:	Margin := $\frac{\text{Limit} - F_{\text{total}}}{F_{\text{total}}}$	Margin	- -0.509	

ENHANCED CAPABILITY EVALUATION:

This enhanced capability evaluation uses the measured overall MOV efficiency to predict the required motor torque during the pressure lock condition (MTrequired). This overall MOV efficiency is determined by dividing the actual motor torque during the latest static test by the measured pullout thrust. This ratio (MOV efficiency) is then used to estimate the required motor torque during pressure lock conditions. The resulting margin must be a minimum of 40% to account for measurement inaccuracies and possible efficiency losses or reduced stem/stem nut coefficient of friction at the estimated pressure lock loads.

$$MT_{required} = F_{total} \cdot \frac{MT_{po}}{F_{po}}$$
$$MT_{avail.} = MR \cdot Tf \left(\frac{DV}{460 \cdot volt}\right)^{n}$$
$$Margin := \frac{MT_{avail.} - MT_{required}}{MT_{required}}$$

MT required = 32.82 •ft lbf

MT avail = $46.423 \cdot ft \cdot lbf$

 $Margin = 41.446 \cdot \%$

The above enhanced capability does not take advantage of the motor locked rotor torque (88.52 ft-lb) In addition, this valve actuates within a few minutes of accident initiation, resulting in little change to motor temperature at the time of opening, thereby, eliminating the temperature degradation factor.

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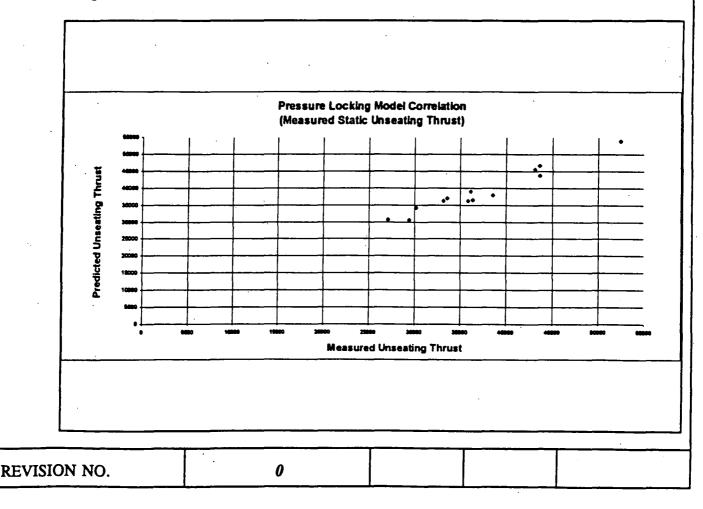
VII. COMPARISON OF MODEL OF OTHER SOURCES OF INFORMATION

Finite Element Calculation to Determine Seat Contact Force due to Bonnet Pressure

Results from the Reference 9 calculation demonstrate that the contact force between the seat and disk which is calculated using the MathCad model above is conservative and reasonably accurate (within 5%) for the 16" valve size modeled in this calculation. Once this Finite Element Analysis calculation is finalized, this calculation will be updated to provide the actual values.

Comparison to Actual Test Data

The reference 10 calculation demonstrates that the MathCad model calculates the pressure locking unseating thrust for the 6" flex-wedge Velan gate valve with good accuracy. In addition, pressure locking testing performed on a 10" Crane 900# Class flex-wedge gate valve at Quad Cities on July 21, 1995 indicates that the model accurately and conservatively predicts that pressure locking unseating force. A graph showing the initial results of this testing is provided below. Further testing is scheduled for later this year. The results of the entire test sequence will be formally documented in a ComEd report after all testing is completed.



	ULATION NO. QDC-1400-M-0080	PROJECT NO. N/A	PAGE NO.1
VIII.	SUMMARY AND CONCLUSIONS		:
	This calculation has determined that the for 69,940 lbf and that the Motor/Gearing Cap Structural Limit for MOV 1-1402-25A is th Assum. 4). The margin was determined by force, in this case the MGC value, and the by the total unseating force to produced a r	ability (MGC) is 33,790 lbf. ne one time value of 83,505 finding the difference betwee unseating force, then dividin	The Open bf (Ref. 12 and en the limiting op
	Based on the conservative MGC calculation enhanced capability evaluation was applied	Ų	
	The calculated margin of 41.5% is greater therefore, a pressure locking event will not function.	•	• •
IX.	ATTACHMENTS		
	Telecopy from Dave Dwyer (Crane-Aloyco	Valves) to Brian Bunte (Con	nEd) dated 5/3/9
•	Page A-1.		
•			· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·
	Page A-1. Telecopy from Bruce Harry (Crane-Aloyco Page A-2.	Valves) to Brian Bunte (Con	· · ·

(FINAL)

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CRANE VALVES Nuclear Operations

FAX TRANSMITTAL

TO BRIAN BUNJIE PHONE FAX 205-66-3-7195 FROM Devid H. Dwyer, Project Engineer PHONE (815) 740 - 7511 FAX (815) 727 - 4248 SUBJECT DISC. DIMENSIONS DATE 5/3/55 TOTAL PAGES / REFFRENCE MESSAGE BRIAN - THE FOLLOWING ARE APPROXIMATE NOMINAL DIMENSIONS FOR THE 10" - 16" VALVES DISCO WE DE DISCUSSED 1485 MORNING. ALL DIMET-SIONS IN INCHES SIZE HUB DIA (D) PLATE THICKMESS (7) MIN MAX 10 2.5 1 3/16 2. 2 10 4,25 2 1/6 16 PLATE THICKNESS 75 GREATEST 17 7000 D 1 **7**7 j RECANDS alere leg ... Att. I.D. <u>A</u>Shit c Caic. No. <u>QX-1400-M</u> Ray. Att. I.D. . Cf. 0080

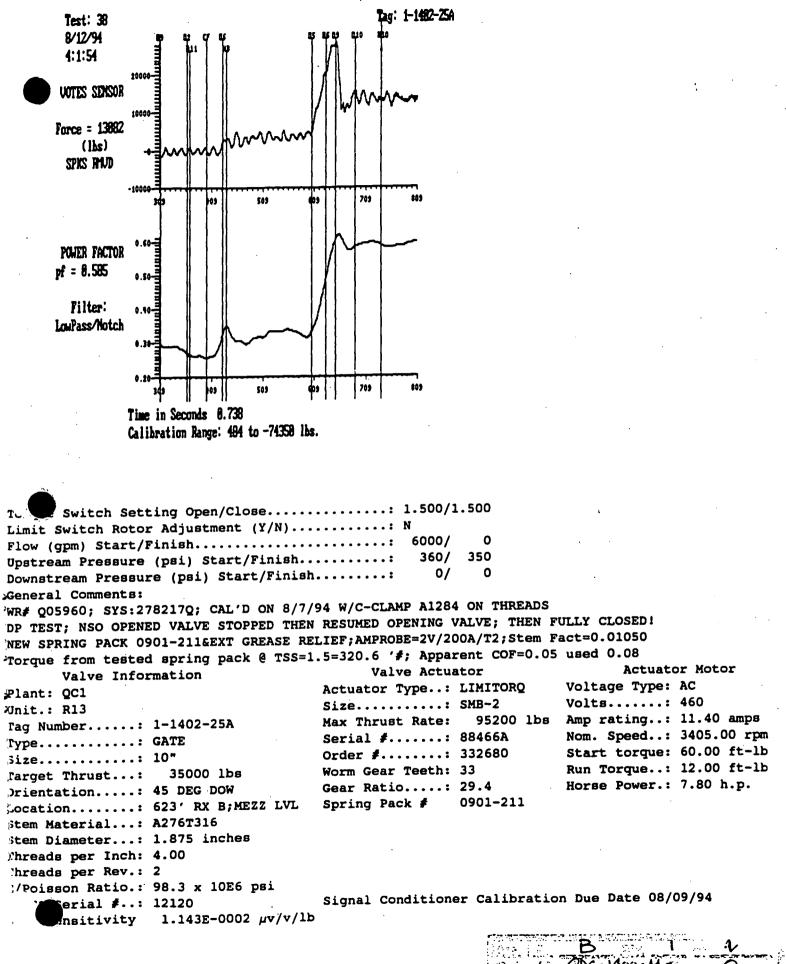
CRANE

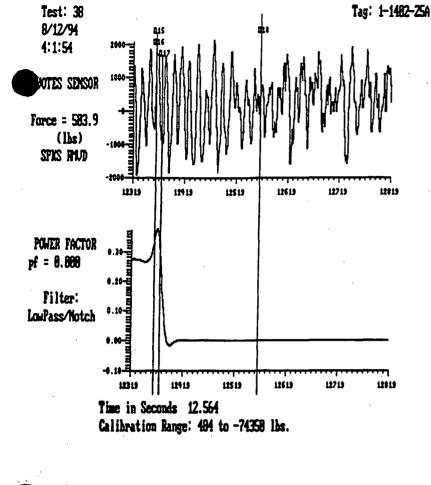
CRANE VALVES NUCLEAR OPERATIONS TELECOPIER TRANSMITTAL 104 North Chicago Street, Joliet, IL 60431

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DATE:	6-16-95	
FO:	BRIAN BUNTE	FROM: BRULE HARRY
ITTLE:	SR. ENG	TITLE: DEV. ENG.
COMPAN	Y:	
PHONE:	708-663-3824	PHONE: <u>8/5-740-7570</u>
FAX:	708-663-7181	FAX: (815) 727-4246
TOTAL P	AGES:	•
SUGJ	ECT! HUB DIM. FO	A 10" AND 16" FIG 783
	FLEXIBLE W	
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		· · ·
THE	LENGTH'S OF THE	FLEXIAIE WEDEF
		FLEXIBLE WEDGE
INTE	mmal Hugs An	LE 1 \$18" AND 2 7/16"
INTE	mmal Hugs An	
INTE	mmal Hugs An	LE 1 \$18" AND 2 7/16"
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INTE	mmal Hugs An	LE 1 \$18" AND 2 7/16"





TC. Switch Setting Open/Close..... 1.500/1.500 imit Switch Rotor Adjustment (Y/N)..... N Flow (gpm) Start/Finish..... 6000/ pstream Pressure (psi) Start/Finish...... 360/ 350 Jownstream Pressure (psi) Start/Finish...... 0/ 0 eneral Comments: **R#** Q05960; SYS:278217Q; CAL'D ON 8/7/94 W/C-CLAMP A1284 ON THREADS P TEST; NSO OPENED VALVE STOPPED THEN RESUMED OPENING VALVE; THEN FULLY CLOSED! JEW SPRING PACK 0901-211&EXT GREASE RELIEF; AMPROBE=2V/200A/T2; Stem Fact=0.01050 'orque from tested spring pack @ TSS=1.5=320.6 '#; Apparent COF=0.05 used 0.08 Valve Information Valve Actuator Actuator Motor lant: QC1 Actuator Type ..: LIMITORQ Voltage Type: AC nit.: R13 Size....: SMB-2 Volts....: 460 Amp rating..: 11.40 amps Max Thrust Rate: 95200 lbs ag Number....: 1-1402-25A Serial #....: 88466A ype..... GATE Nom. Speed..: 3405.00 rpm Order #....: 332680 ize....: 10" Start torque: 60.00 ft-lb arget Thrust...: 35000 lbs Worm Gear Teeth: 33 Run Torque..: 12.00 ft-1b rientation....: 45 DEG DOW Gear Ratio....: 29.4 Horse Power.: 7.80 h.p. Spring Pack # pcation.....: 623' RX B; MEZZ LVL 0901-211 tem Material...: A276T316 tem Diameter...: 1.875 inches hreads per Inch: 4.00 hreads per Rev.: 2 /Poisson Ratio.: 98.3 x 10E6 psi Signal Conditioner Calibration Due Date 08/09/94 rial #..: 12120 1.143E-0002 $\mu v/v/lb$ sitivity

Ait. I.D. B Sht Z of 2 Cale, No. 92-140-M Trav 0