

In the Matter of:

Entergy Nuclear Operations, Inc.
(Indian Point Nuclear Generating Units 2 and 3)

ASLBP #: 07-858-03-LR-BD01
Docket #: 05000247 | 05000286
Exhibit #: RIV00053J-00-BD01
Admitted: 10/15/2012
Rejected:
Other:

Identified: 10/15/2012
Withdrawn:
Stricken:

RIV00053J

Submitted: December 27, 2011

COMBUSTION ENGINEERING, INC.

ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

CHARGE NO. _____

DESCRIPTION: STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL WALL TRANSITION

NUMBER 5-202-P | A 219

SHEET 11 OF 26

DATE MAY 26, 1966 BY COOPER

CHECK DATE MAY 26, 1966 BY ALEXANDER

5- DETAILED ANALYSIS:

e. STRESSES:

THE FOLLOWING EXPRESSIONS WILL BE USED TO CALCULATE STRESSES AT THE FOUR LOCATIONS AS SHOWN ON SHEET 4

LOCATION 1:

$$\sigma_x = \frac{6M_1}{t_1^2} + \frac{b_1^2 P}{2R_1 t_1} + \frac{E \Delta_1 (T_m - T)}{1 - \nu} = \underline{0.05192 M_1 + 3.72970 P + 0.30714 (T_m - T)}$$

$$\sigma_\theta = \frac{\sqrt{6} M_1}{t_1} + \frac{E \Delta_1}{R_1} + \frac{b_1 P}{t_1} + \frac{E \Delta_1 (T_m - T)}{1 - \nu} = \underline{0.01558 M_1 + 0.01099 E \Delta_1 + 7.94772 P + 0.30714 (T_m - T)}$$

LOCATION 2:

$$\sigma_x = -\frac{6M_2}{t_2^2} + \frac{b_2^2 P}{2R_2 t_2} + \frac{E \Delta_2 (T_m - T)}{1 - \nu} = \underline{-0.05192 M_2 + 3.72970 P + 0.30714 (T_m - T)}$$

$$\sigma_\theta = -\frac{\sqrt{6} M_2}{t_2} + \frac{E \Delta_2}{R_2} + \frac{b_2 P}{t_2} + \frac{E \Delta_2 (T_m - T)}{1 - \nu} = \underline{-0.01558 M_2 + 0.01099 E \Delta_2 + 7.94772 P + 0.30714 (T_m - T)}$$

LOCATION 3:

$$\sigma_x = \frac{6M_3}{t_3^2} + \frac{b_3^2 P}{2R_3 t_3} + \frac{E \Delta_3 (T_m - T)}{1 - \nu} = \underline{0.08066 M_3 + 4.76490 P + 0.30714 (T_m - T)}$$

$$\sigma_\theta = \frac{\sqrt{6} M_3}{t_3} + \frac{E \Delta_3}{R_3} + \frac{b_3 P}{t_3} + \frac{E \Delta_3 (T_m - T)}{1 - \nu} = \underline{0.02420 M_3 + 0.01099 E \Delta_3 + 10.02899 P + 0.30714 (T_m - T)}$$

LOCATION 4:

$$\sigma_x = -\frac{6M_4}{t_4^2} + \frac{b_4^2 P}{2R_4 t_4} + \frac{E \Delta_4 (T_m - T)}{1 - \nu} = \underline{-0.08066 M_4 + 4.76490 P + 0.30714 (T_m - T)}$$

$$\sigma_\theta = -\frac{\sqrt{6} M_4}{t_4} + \frac{E \Delta_4}{R_4} + \frac{b_4 P}{t_4} + \frac{E \Delta_4 (T_m - T)}{1 - \nu} = \underline{-0.02420 M_4 + 0.01099 E \Delta_4 + 10.02899 P + 0.30714 (T_m - T)}$$

Note: T_m = Thermal Stress Will Be Conservatively Treated As A SHIM TYPE STRESS.

$E = 0.215$ FOR SA-302B MATERIAL AT 550°F

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-D | A 24
SHEET 12 OF 24
DATE MAY 26, 1966 BY LEWIS
CHECK DATE MAY 26, 1966 BY ALSTON

CHARGE NO. _____
DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITION

5. DETAILED ANALYSIS:
C. STRESSES:

Location	M	$\pm \frac{6M}{t^2}$	$\frac{PR}{2t}$	σ_r	$\pm \frac{2EM}{t^2}$	EA	$\frac{EA}{R}$	$\frac{PR}{t}$	σ_o	σ_r	STRESS INTENSITY		
											$\sigma_o - \sigma_r$	$\sigma_r - \sigma_r$	$\sigma_o - \sigma_r$
1	13.068	0.68	9.32	0.00	0.20	163.46	1.80	19.87	21.87	-2.5	-1.87	125.0	24.57
2	13.068	-0.68	9.32	8.64	-0.20	163.46	1.80	19.87	21.47	0	-12.83	8.64	21.47
3	-0.324	-0.03	11.91	11.88	-0.01	-186.97	-2.05	25.07	23.01	-2.5	-11.13	14.38	25.51
4	-0.324	0.03	11.91	11.94	0.01	-186.97	-2.05	25.07	23.03	0	-11.09	11.94	23.03

THE VALUES OF H & M ARE TAKEN FROM SHEET 10.
THE MOMENT EQUATIONS ARE GIVEN ON SHEETS 7 & 9.

CRITERION 5.C.1 - PRIMARY GENERAL MEMBRANE:

$S.I._{max} = \sigma_o - \sigma_r = \frac{PR}{t} - \left(-\frac{P}{2}\right) = 25.07 + 1.25 = \underline{26.3 \text{ ksi}} < 26.7 \text{ ksi}$
@ LOCATION 3 & 4

CRITERION 5.C.2 - LOCAL MEMBRANE STRESS:

$S.I. = \sigma_o = \frac{EA}{R} = -2.05 < 40 \text{ ksi}$ @ LOCATION 3 & 4

OR COMBINED WITH 1. ABOVE,

$S.I._{max} = \underline{24.3 \text{ ksi}} < 40 \text{ ksi}$ @ LOCATION 3 & 4

CRITERION 5.C.4 - RANGE OF STRESS INTENSITY:

$(S.I.)_{max} = (\sigma_o - \sigma_r) + \left[\frac{PR}{t} + \frac{EA}{R} + \frac{2EM}{t^2} \right] + P = \underline{28.0 \text{ ksi}} < 80 \text{ ksi}$ @ LOCATION 3

COMBUSTION ENGINEERING, INC.
 ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 6-202-P-1-A-200
 SHEET 13 OF 24

CHARGE NO. _____

DATE MAY 26, 1966 BY W. L. GARDNER

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL WALL TRANSITION

CHECK DATE MAY 26, 1966 BY W. L. GARDNER

5- DETAILED ANALYSIS:
C- STRESSES:

THE FOLLOWING TABLES GIVE THE COMBINED PRESSURE AND THERMAL STRESSES (NEGLECTING STRESS CONCENTRATION FACTORS).

LOCATION - 1

TRANSIENT	INTERNAL PRESSURE KSI/A	(T _m -T) °F	THERMAL STRESS F _{T=60}	PRESSURE STRESS			TOTAL STRESS			STRESS INTENSITY		
				F _r	F _o	F _r	F _r	F _o	F _r	F _{r-60}	F _{r-6r}	F _{o-6r}
Steady State	2.250	0	0	9.00	19.68	-2.25	9.00	19.68	-2.25	-10.68	11.25	21.93
a 4.47 hrs	2.250	-65	-19.96	9.00	19.68	-2.25	-10.96	-0.28	2.25	-10.68	-8.71	1.97
b 4.47 hrs	0.315	65	19.96	1.26	2.76	-0.32	21.22	22.72	-0.32	-1.50	21.54	23.04
c 20 min	2.250	-7.8	-2.40	9.00	19.68	-2.25	6.60	17.28	-2.25	-10.68	8.95	19.53
d 20 min	2.250	7.8	2.40	9.00	19.68	-2.25	11.40	22.88	-2.25	-10.68	13.65	24.33
e 100 sec	2.160	11.2	3.44	8.56	18.72	-2.14	12.00	22.16	-2.14	-10.16	18.14	20.30
225 sec	2.275	1.7	0.52	9.10	19.90	-2.28	9.62	20.42	-2.28	-10.80	11.90	22.70
f 40 sec	2.320	-9.3	-2.86	9.28	20.29	-2.32	6.42	17.43	-2.32	-11.01	8.74	19.75
100 sec	2.260	-13.3	-4.08	9.04	19.77	-2.26	4.46	15.69	-2.26	-10.73	7.22	17.95
260 sec	2.160	-1.3	-0.40	8.56	18.72	-2.14	9.16	18.32	-2.14	-10.16	10.30	20.46
g 2 min	2.370	-12.0	-3.49	9.48	20.73	-2.37	5.79	17.06	-2.37	-11.25	8.16	19.41
32 min	2.350	-15.0	-4.61	9.60	20.53	-2.35	6.79	15.92	-2.35	-11.13	7.18	18.27
10.4 min	2.150	0	0	8.60	18.81	-2.15	8.60	18.81	-2.15	-10.21	10.75	20.96
h 10 sec	2.220	-9.5	-2.92	8.88	19.42	-2.22	5.96	16.50	-2.22	-10.84	9.18	18.72
65 sec	1.910	3.5	2.61	7.64	16.71	-1.91	10.29	19.32	-1.91	-9.07	12.16	21.23
i 220 min	3.12	0	0	12.50	27.36	-3.13	12.50	27.36	-3.13	-14.84	15.63	30.43
j 3.5 hrs	1.250	-6.2	-19.66	5.00	10.94	-1.25	-14.66	-0.72	-1.25	-5.94	13.41	-7.47
5.5 hrs	2.300	0	0	10.00	21.87	-2.50	10.00	21.87	-2.50	-11.87	12.50	20.37
3.5 hrs	0.315	64	19.66	1.26	2.76	-0.32	20.92	22.92	-0.32	-1.50	21.24	22.74
k ~	2.250	6.0	1.84	9.40	20.53	-2.25	11.26	22.37	-2.25	-11.13	13.59	24.72
~	2.150	-6.0	-1.84	8.60	18.81	-2.15	6.76	16.97	-2.15	-10.21	9.91	19.12
l 12 sec	2.250	33.3	10.23	9.00	19.68	-2.25	19.23	29.91	-2.25	-10.68	21.48	32.16
m 16 sec	2.760	-32.2	-9.28	11.04	24.14	-2.76	1.76	14.06	-2.76	-16.10	4.82	17.64
28 sec	2.110	-41.2	-12.65	8.48	18.53	-2.12	-4.17	5.96	-2.12	-10.87	-2.05	8.02
160 sec	1.440	4.8	1.47	5.76	12.60	-1.44	7.33	14.07	-1.44	-6.84	8.67	15.51
n 33 sec	1.300	117	55.94	1.20	2.62	-1.30	37.14	38.56	-0.80	-1.42	57.44	38.86
5.4 sec	0.700	197	60.51	2.80	6.12	-0.70	63.31	66.63	-0.70	-2.32	64.01	67.33

$\sigma_{max} = (F_o - F_r) = 37.9 \text{ ksi} < 3S_m = 84.1 \text{ ksi}$ (CRITERION 5-C-4)

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 6-201-P 1A
SHEET 14 OF 24
DATE May 26, 1966 BY W.C. HARRIS
CHECK DATE May 26, 1966 BY W.C. HARRIS

CHARGE NO. _____

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITION

5- DETAILED ANALYSIS:
C. STRESSES:

LOCATION - 2

STRESS STATE	INTERNAL PRESSURE KSI	(T _m -T) °F	THERMAL STRESS T _m -T _c	PRESSURE STRESS			TOTAL STRESS			STRESS INTENSITY		
				F _r	F _o	F _t	F _r	F _o	F _t	F _{r}-F_o}	F _{o}-F_t}	F _{t}-F_r}
Standard State	2.250	0	0	7.78	19.32	0	7.78	19.32	0	-11.54	7.78	19.32
a	4.17 hrs	2.250	33	10.14	7.78	19.32	17.92	29.46	0	-11.54	17.92	29.46
b	4.17 hrs	0.315	-33	-10.14	1.09	2.70	-9.05	-7.44		-1.61	-9.05	-7.44
c	20 min	2.250	0	0	7.78	19.32	7.78	19.32		-11.54	7.78	19.32
d	20 min	2.250			7.78	19.32	7.78	19.32		-11.54	7.78	19.32
e	100 sec	2.140			7.40	18.37	7.40	18.37		-10.97	7.40	18.37
	225 sec	2.275			7.07	19.53	7.07	19.53		-11.66	7.87	19.53
f	40 sec	2.320			8.02	19.92	8.02	19.92		-11.90	8.02	19.92
	100 sec	2.260			7.82	19.40	7.82	19.40		-11.58	7.82	19.40
	260 sec	2.140			7.40	18.37	7.40	18.37		-10.97	7.40	18.37
g	2 min	2.370			8.20	20.35	8.20	20.35		-12.15	8.20	20.35
	32 min	2.350			8.13	20.17	8.13	20.17		-12.04	8.13	20.17
	10.4 min	2.150			7.44	18.46	7.44	18.46		-11.02	7.44	18.46
h	10 sec	2.220			7.68	19.06	7.68	19.06		-11.38	7.68	19.06
	65 sec	1.910			6.61	16.40	6.61	16.40		-9.79	6.61	16.40
i	220 min	3.125	↓	↓	10.81	26.83	10.81	26.83		-16.02	10.81	26.83
j	3.5 hrs	1.250	33	10.14	4.52	10.73	14.46	20.87		-6.41	14.46	20.87
	5.5 hrs	2.800	0	0	8.65	21.46	8.65	21.46		-12.81	8.65	21.46
	3.5 hrs	0.315	-33	-10.14	1.09	2.70	-9.05	-7.44		-1.61	-9.05	-7.44
k	~	2.350	0	0	8.13	20.17	8.13	20.17		-12.04	8.13	20.17
	~	2.150			7.44	18.46	7.44	18.46		-11.02	7.44	18.46
l	12 sec	2.250			7.78	19.32	7.78	19.32		-11.54	7.78	19.32
	10 sec	2.700			9.54	23.69	9.54	23.69		-14.15	9.54	23.69
m	20 sec	2.120			7.33	18.20	7.33	18.20		-10.87	7.33	18.20
	100 sec	1.440			4.98	12.36	4.98	12.36		-7.38	4.98	12.36
n	33 sec	1.500			1.04	2.58	1.04	2.58		-1.54	1.04	2.58
	54 sec	0.700			2.42	6.01	2.42	6.01		-3.59	2.42	6.01

$S.I._{max} = (F_o - F_r) = 36.9 \text{ ksi} < SS_m = 80.1 \text{ ksi}$ (CRITERION 5C-4)

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-201-P | A 289

SHEET 15 OF 24

CHARGE NO. _____

DATE MAY 26, 1966 BY COOPER

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
TILE VESSEL WALL TRANSITION

CHECK DATE MAY 26, 1966 BY ALEXANDER

5- DETAILED ANALYSIS:

C. STRESSES:

LOCATION - 3

TRANSIENT	INTERNAL PRESSURE KPSIA	(T _{in} -T) °F	THERMAL STRESS σ _t -σ _b	PRESSURE STRESS			TOTAL STRESS			STRESS INTENSITY			
				σ _t	σ _b	σ _r	σ _t	σ _b	σ _r	σ _t -σ _b	σ _t -σ _r	σ _b -σ _r	
Steady State	2.250	0	0	10.70	20.71	-2.25	10.70	20.71	-2.25	-10.01	12.95	22.96	
a 4.47 hrs	2.250	-43	-13.21	10.70	20.71	-2.25	-2.51	7.50	-2.25	-10.01	-0.26	9.75	
b 6.47 hrs	0.315	43	13.21	1.50	2.90	-0.32	14.71	16.11	-0.32	-1.40	15.03	16.43	
c 20 min	2.250	-7.8	-2.40	10.70	20.71	-2.25	8.30	18.31	-2.25	-10.01	10.55	20.56	
d 20 min	2.250	7.8	2.40	10.70	20.71	-2.25	13.10	23.11	-2.25	-10.01	15.35	25.36	
e	100 sec	2.140	11.2	3.44	10.17	19.70	-2.14	13.61	23.16	-2.14	-9.53	15.75	25.29
	225 sec	2.275	1.7	5.52	10.82	20.94	-2.20	11.34	21.46	-2.20	-10.12	15.62	23.72
f	40 sec	2.370	-9.3	-2.86	11.03	21.85	-2.32	9.17	18.49	-2.32	-10.32	10.49	20.81
	100 sec	2.260	-13.3	-4.09	10.75	20.80	-2.26	6.67	16.72	-2.26	-10.05	8.93	18.93
	260 sec	2.140	-1.7	-0.40	10.17	19.70	-2.14	9.77	19.30	-2.14	-9.53	11.41	21.44
g	2 min	2.370	-12.0	-3.69	11.27	21.81	-2.37	7.58	18.12	-2.37	-10.54	9.95	20.49
	3.2 min	2.350	-15.0	-4.61	11.17	21.63	-2.35	6.56	17.02	-2.35	-10.46	8.91	19.37
	10.4 min	2.150	0	0	10.22	19.79	-2.15	10.22	19.79	-2.15	-9.57	12.37	21.94
h	10 sec	2.220	-9.5	-2.92	10.55	20.43	-2.22	7.63	17.51	-2.22	-9.88	9.95	19.73
	65 sec	1.910	8.5	2.61	9.08	17.58	-1.91	11.69	20.19	-1.91	-8.50	13.60	22.10
i 220 min	3.125	0	0	14.86	28.76	-3.13	14.86	28.76	-3.13	-13.90	17.99	31.89	
j	3.5 hrs	1.250	-43	-13.21	5.94	11.50	-1.25	-7.27	-1.71	-1.25	-5.56	-6.02	-0.46
	5.5 hrs	2.500	0	0	11.89	23.01	-2.50	11.89	23.01	-2.50	-11.12	14.39	25.51
	3.5 hrs	0.315	43	13.21	1.50	2.90	-0.32	14.71	16.11	-0.32	-1.40	15.03	16.43
k	~	2.350	6.0	1.84	11.17	21.63	-2.35	13.01	23.07	-2.35	-10.46	15.36	25.82
	~	2.150	-6.0	-1.84	10.22	19.79	-2.15	8.38	17.95	-2.15	-9.57	10.53	20.10
l 12 sec	2.250	33.3	10.23	10.70	20.71	-2.25	20.93	30.94	-2.25	-10.01	22.08	33.19	
m	10 sec	2.760	-30.2	-9.28	13.12	25.80	-2.76	3.94	16.12	-2.76	-12.28	6.40	18.18
	28 sec	2.120	-41.2	-12.65	10.08	19.91	-2.12	-2.57	6.84	-2.12	-9.43	-0.45	9.98
	160 sec	1.460	4.8	1.47	6.85	13.25	-1.46	0.32	14.72	-1.46	-6.40	9.76	16.16
n	33 sec	1.300	117	35.94	1.43	2.76	-0.30	97.37	38.70	-0.30	-1.33	37.67	59.00
	5.4 sec	0.700	197	60.51	3.33	6.44	-0.70	63.84	66.95	-0.70	-3.11	64.54	67.65

S.I._{max} = (σ_t - σ_r) = 32.4 ksi < 3S_m = 80.1 ksi. CRITERION 5-C-4

COMBUSTION ENGINEERING, INC.
 ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER E-202-P | A 204
 SHEET 16 OF 24
 DATE MAY 26, 1966 BY CSH/REDA
 CHECK DATE MAY 26, 1966 BY ALEXANDER

CHARGE NO. _____
 DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITION

5. DETAILED ANALYSIS:
C. STRESSES:

LOCATION - 4

TRANSIENT	INTERNAL PRESSURE KSI	(T _m -T) °F	THERMAL STRESS F _t -F _o	PRESSURE STRESS			TOTAL STRESS			STRESS INTENSITY		
				F _x	F _o	F _r	F _x	F _o	F _r	F _x -F _o	F _x -F _r	F _o -F _r
Steady State	2.250	0	0	10.74	20.72	0	10.74	20.72	0	-9.98	10.74	20.72
a	4.47 hrs	2.250	22	6.76	10.74	20.72	17.50	27.48	0	-9.98	17.50	27.48
b	4.47 hrs	0.315	-22	-6.76	1.50	2.90	-5.26	-3.86	0	-1.60	-5.26	-3.86
c	20 min	2.250	0	0	10.74	20.72	10.74	20.72	0	-9.98	10.74	20.72
d	20 min	2.250			10.74	20.72	10.74	20.72		-9.98	10.74	20.72
e	100 sec	2.140			10.22	19.71	10.22	19.71		-9.99	10.22	19.71
	225 sec	2.275			10.88	20.95	10.88	20.95		-10.07	10.88	20.95
f	40 sec	2.320			11.08	21.37	11.08	21.37		-10.29	11.08	21.37
	110 sec	2.260			10.79	20.82	10.79	20.82		-10.03	10.79	20.82
	260 sec	2.140			10.22	19.71	10.22	19.71		-9.99	10.22	19.71
g	2 min	2.370			11.32	21.83	11.32	21.83		-10.51	11.32	21.83
	3.2 min	2.350			11.22	21.64	11.22	21.64		-10.42	11.22	21.64
	10.4 min	2.150			10.27	19.80	10.27	19.80		-9.53	10.27	19.80
h	10 sec	2.220			10.60	20.45	10.60	20.45		-9.85	10.60	20.45
	65 sec	1.910			9.12	17.99	9.12	17.99		-9.47	9.12	17.99
i	220 min	3.125			14.92	28.78	14.92	28.78		-13.86	14.92	28.78
j	3.5 hrs	1.250	22	6.76	5.97	11.51	12.73	18.27		-5.54	12.73	18.27
	5.5	2.500	0	0	11.94	23.02	11.94	23.02		-11.08	11.94	23.02
	35 hrs	0.315	-22	-6.76	1.50	2.90	-5.26	-3.86	0	-1.60	-5.26	-3.86
k	~	2.350	0	0	11.22	21.64	11.22	21.64		-10.42	11.22	21.64
	~	2.150			10.27	19.80	10.27	19.80		-9.53	10.27	19.80
l	12 sec	2.250			10.74	20.72	10.74	20.72		-9.98	10.74	20.72
m	10 sec	2.760			12.18	25.42	12.18	25.42		-12.24	12.18	25.42
	20 sec	2.120			10.12	19.53	10.12	19.53		-9.41	10.12	19.53
	160 sec	1.440			6.88	13.26	6.88	13.26		-6.30	6.88	13.26
n	33 sec	0.300			1.43	2.76	1.43	2.76		-1.33	1.43	2.76
	54 sec	0.700			3.34	6.45	3.34	6.45		-3.11	3.34	6.45

S.I._{max} = F_o - F_r = 32.6 ksi < 3S_m = 90.1 ksi (CRITERION 5-C-4)

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER S-302-P | A 289
SHEET 17 OF 24

CHARGE NO. _____

DATE MAY 26, 1966 BY LOTT

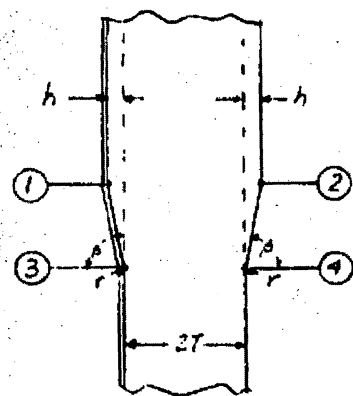
DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL WALL TRANSITION

CHECK DATE MAY 26, 1966 BY DESINGER

5. DETAILED ANALYSIS:

4. FATIGUE EVALUATION:

IN ORDER TO PERFORM THE FATIGUE EVALUATION, PEAK STRESSES MUST BE KNOWN AT THE FOUR LOCATIONS AS SHOWN BELOW. THE STRESS EXPRESSIONS GIVEN ON SHEET 10 WILL BE MODIFIED TO ACCOUNT FOR STRESS CONCENTRATIONS. WITH THE PEAK STRESSES, A FATIGUE EVALUATION WILL BE MADE BY THE CUMULATIVE METHOD WHERE SUPERPOSITION OF ALL CYCLES IS TAKEN INTO CONSIDERATION.



BY INSPECTION, WE SEE THAT THE STRESS CONCENTRATION FACTORS AT LOCATIONS 1 & 2 EQUAL 1. AT LOCATIONS 3 & 4, STRESS CONCENTRATION FACTOR FOR BENDING AND TENSION WILL BE DETERMINED BY THE METHOD PRESENTED IN REFERENCE 3.

$T = 4.3125"$
 $h = 1.063"$
 $\beta = 80"$
 $r = 2"$

$\frac{r}{T} = 0.464$
 $\frac{r}{h} = 1.891$
 $\frac{r}{r_0} = 0.999$

FROM FIGURE A.7-1 OF REF. 3

FROM FIGURE A.7-2 OF REF. 3

$K_T = 1.85$
 $K_B = 1.53$

$\left[\frac{K' - 1}{K_B - 1} \right] = 0.4$

$K_T' = 1 + 0.4(K_T - 1) = \underline{1.34}$

$K_B' = 1 + 0.4(K_B - 1) = \underline{1.21}$

LOCATION 3

$\sigma_x = \frac{6M}{t^2} K_B' + \frac{b_2 P}{2r_1 t_1} K_T' + \frac{E\alpha(T_m - T)}{1 - \nu} K_T' = \underline{6.37232P + 0.41157(T_m - T)}$

$\sigma_\theta = \frac{3b_1 M}{t^2} K_B' + \frac{E\alpha t_1}{r_1} + \frac{b_2 P}{t_2} + \frac{E\alpha(T_m - T)}{1 - \nu} = \underline{9.20326P + 0.30714(T_m - T)}$

LOCATION 4

$\sigma_x = \frac{6M}{t^2} K_B' + \frac{b_2 P}{2r_1 t_1} K_T' + \frac{E\alpha(T_m - T)}{1 - \nu} K_T' = \underline{6.39762P + 0.41157(T_m - T)}$

$\sigma_\theta = \frac{3b_1 M}{t^2} K_B' + \frac{E\alpha t_1}{r_1} + \frac{b_2 P}{t_2} + \frac{E\alpha(T_m - T)}{1 - \nu} = \underline{9.21096P + 0.30714(T_m - T)}$

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-P | A 200

SHEET 10 OF 24

CHARGE NO. _____

DATE May 26, 1966 BY CHERRY

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITIONS

CHECK DATE May 26, 1966 BY ALVIN

S- DETAILED ANALYSIS:

F- FATIGUE EVALUATION:

PEAK STRESS A- LOCATION -3

TRANSIENT	INTERNAL PRESSURE PSIA	T _m -T °F	THERMAL STRESS		PRESSURE STRESS			PEAK STRESS		
			F _t	F _b	F _t	F _b	F _r	F _t	F _b	F _r
Steady State	2.250	0	0	0	14.34	20.71	-2.25	14.34	20.71	-2.25
a 4.47hrs	2.250	-43	-17.70	-13.21	14.34	20.71	-2.25	-3.36	7.50	-2.25
b 4.47hrs	0.315	43	17.70	13.21	2.01	2.90	-0.32	19.71	16.11	-0.32
c 20 min	2.250	-7.8	-3.21	-2.40	14.34	20.71	-2.25	11.13	18.31	-2.25
d 20 min	2.250	7.8	3.21	2.40	14.34	20.71	-2.25	17.55	23.11	-2.25
e 10 SEC	2.140	11.2	4.61	3.44	13.64	19.69	-2.14	19.25	23.13	-2.14
225 SEC	2.275	1.7	0.70	0.52	14.50	20.94	-2.28	15.20	21.46	-2.28
f 40 SEC	2.320	-9.3	-3.83	-2.86	14.78	21.35	-2.32	10.95	18.49	-2.32
100 SEC	2.260	-13.3	-5.47	-4.08	14.60	20.80	-2.26	8.93	16.72	-2.26
260 SEC	2.140	-1.3	-0.54	-0.40	13.64	19.69	-2.14	13.10	19.29	-2.14
g 2 min	2.370	-12.0	-4.94	-3.69	15.10	21.81	-2.37	10.16	18.12	-2.37
3.2 min	2.350	-5.0	-6.17	-4.61	14.97	21.63	-2.35	8.80	17.02	-2.35
10.4 min	2.150	0	0	0	13.70	19.79	-2.15	13.70	19.79	-2.15
h 10 SEC	2.220	-9.5	-3.91	-2.92	14.15	20.43	-2.22	10.24	17.51	-2.22
65 SEC	1.910	8.5	3.50	2.61	12.17	17.58	-1.91	15.67	20.19	-1.91
i 220 min	3.125	0	0	0	19.91	29.76	-3.13	19.91	29.76	-3.13
Startup	1.250	-43	-17.70	-13.21	7.96	11.50	-1.25	-9.74	-7.71	-1.25
j S.S. (normal)	2.600	0	0	0	15.93	23.01	-2.50	15.93	23.01	-2.50
3.5 hrs	0.315	43	17.70	13.21	2.01	2.90	-0.32	19.71	16.11	-0.32
k ~	2.350	6.0	2.47	1.84	14.97	21.63	-2.35	17.44	23.47	-2.35
~	2.150	-6.0	-2.47	-1.84	13.70	19.79	-2.15	11.23	17.95	-2.15
l 12 SEC	2.250	33.3	13.71	10.23	14.34	20.71	-2.25	20.05	30.94	-2.25
m 10 SEC	2.760	-30.2	-12.43	-9.28	17.59	25.40	-2.76	5.16	16.12	-2.76
20 SEC	2.120	-41.2	-16.96	-12.65	13.51	19.51	-2.12	-3.45	6.86	-2.12
160 SEC	1.440	4.8	1.98	1.47	9.18	13.25	-1.44	11.16	14.72	-1.44
n 55 SEC	3.800	117	48.15	35.94	1.91	2.76	-0.30	50.06	38.70	-0.30
50 SEC	0.700	197	81.08	60.51	4.46	6.44	-0.70	85.54	66.95	-0.70

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER S-202-P | A26
SHEET 19 OF 24
DATE MAY 26, 1966 BY GC
CHECK DATE MAY 26, 1966 BY ALL

CHARGE NO. _____
DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITION

S- DETAILED ANALYSIS:
1. FATIGUE EVALUATION:

PEAK STRESS AT LOCATION - 4

TRANSIENT	INTERNAL PRESSURE KSI	T _m -T °F	TANGENTIAL STRESS		PRESSURE STRESS			PEAK STRESS			
			F _t	F _o	F _r	F _o	F _r	F _t	F _o	F _r	
Steady State	2.250	0	0	0	14.39	20.72	0	14.39	20.72	0	
a 4.47 hrs	2.250	22	9.05	6.76	14.39	20.72		23.44	27.49		
b 4.47 hrs	0.315	-22	-9.05	-6.76	2.01	2.90		-7.04	-3.86		
c 20 min	2.250	0	0	0	14.39	20.72		14.39	20.72		
d 20 min	2.250				14.39	20.72		14.39	20.72		
e	10 SEC	2.100			13.69	19.71		13.69	19.71		
	225 SEC	2.275			14.55	20.95		14.55	20.95		
f	40 SEC	2.320			14.84	21.97		14.84	21.97		
	100 SEC	2.260			14.46	20.82		14.46	20.82		
	240 SEC	2.140			13.69	19.71		13.69	19.71		
g	2 min	2.370			15.16	21.83		15.16	21.83		
	3.2 min	2.350			15.03	21.65		15.03	21.65		
	10.8 min	2.150			13.75	19.80		13.75	19.80		
h	10 SEC	2.220			14.20	20.45		14.20	20.45		
	65 SEC	1.910			12.22	17.59		12.22	17.59		
i	220 min	3.125			19.99	28.78		19.99	28.78		
j	Start 3.5 hrs	1.250	22	9.05	6.76	8.00	11.51		17.05	18.37	
	5.5 ignoma	2.500	0	0	0	15.99	23.02		15.99	23.02	
	35 hrs	0.315	-22	-9.05	-6.76	2.01	2.90		-7.04	-3.86	
k	~	2.350	0	0	0	15.03	21.65		15.03	21.65	
	~	2.150				13.75	19.80		13.75	19.80	
l	12 SEC	2.250				14.39	20.72		14.39	20.72	
m	10 sec	2.760				17.66	25.42		17.66	25.42	
	20 sec	2.120				13.56	19.53		13.56	19.53	
	160 sec	1.440				9.21	13.26		9.21	13.26	
n	33 sec	1.300				1.92	2.76		1.92	2.76	
	54 sec	0.700				4.48	6.45		4.48	6.45	

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-P | A 208

SHEET 20 OF 26

CHARGE NO. _____

DATE MAY 26, 1966 BY GRAND

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
TIRE VESSEL WALL TRANSITION

CHECK DATE MAY 26, 1966 BY GRAND

5 DETAILED ANALYSIS:
1. FATIGUE EVALUATION:

PEAK STRESS INTENSITIES

TRANSIENT	LOCATION-1			LOCATION-2			LOCATION-3			LOCATION-4		
	$\sigma_x - \sigma_y$	$\sigma_x - \sigma_z$	$\sigma_y - \sigma_z$	$\sigma_x - \sigma_y$	$\sigma_x - \sigma_z$	$\sigma_y - \sigma_z$	$\sigma_x - \sigma_y$	$\sigma_x - \sigma_z$	$\sigma_y - \sigma_z$	$\sigma_x - \sigma_y$	$\sigma_x - \sigma_z$	$\sigma_y - \sigma_z$
STEADY STATE	-10.60	11.25	21.93	-11.54	7.70	19.32	-6.37	16.59	22.96	-6.33	14.39	20.72
a 4.47 hrs	-10.03	-9.71	1.97	-11.54	17.92	29.46	-10.86	-1.11	9.75	-4.04	23.44	27.48
b 4.47 hrs	-1.50	21.54	23.04	-1.61	-9.05	-7.44	3.60	20.03	16.43	-3.18	-7.04	-3.86
c 20 min	-10.60	8.95	19.53	-11.54	7.78	19.32	-7.18	13.30	20.56	-6.33	16.59	20.72
d 20 min	-10.69	13.65	24.33	-11.54	7.78	19.32	-5.56	19.80	25.36	-6.33	14.39	20.72
e 100 sec	-10.16	14.14	24.20	-10.97	7.40	18.37	-4.88	20.39	25.27	-6.02	13.69	19.71
225 sec	-10.80	11.90	22.70	-11.66	7.87	19.53	-6.26	17.88	22.76	-6.40	14.55	20.95
f 40 sec	-11.01	8.74	19.75	-11.90	8.02	19.92	-7.54	13.27	20.81	-6.53	14.84	21.37
100 sec	-10.73	7.22	17.95	-11.58	7.82	19.40	-7.79	11.19	18.98	-6.36	14.46	20.22
260 sec	-10.16	10.30	20.46	-10.97	7.40	18.37	-6.19	15.24	21.43	-6.02	13.69	19.71
g 2 min	-11.25	8.16	19.41	-12.15	8.20	20.35	-7.96	12.53	20.49	-6.67	15.16	21.83
3.2 min	-11.13	7.14	18.27	-12.04	8.13	20.17	-8.22	11.15	19.37	-6.62	15.03	21.65
10.4 min	-10.21	10.75	20.96	-11.02	7.44	18.46	-6.09	15.85	21.94	-6.05	13.75	19.80
h 10 sec	-10.54	8.18	18.72	-11.38	7.68	19.46	-7.27	12.46	19.73	-6.25	14.20	20.45
65 sec	-9.07	12.16	21.23	-9.79	6.61	16.40	-4.52	17.58	22.10	-5.37	12.22	17.59
i 220 min	-14.84	15.63	30.43	-16.02	10.81	26.83	-8.95	23.04	31.89	-9.79	19.99	29.78
16000 3.5 hrs	-5.94	13.41	-7.47	-6.41	14.46	20.87	-8.03	-8.49	-0.46	-1.22	17.05	18.27
5.5 3.5 hrs	-11.87	12.50	24.37	-12.81	8.65	21.46	-7.08	18.43	25.51	-7.03	15.99	23.02
3.5 hrs	-1.50	21.24	22.76	-1.61	-9.05	-7.44	3.60	20.03	16.43	-3.18	-7.04	-3.86
k ~	-11.13	13.59	24.72	-12.04	8.13	20.17	-6.03	19.79	25.82	-6.62	15.03	21.65
~	-14.21	8.91	19.12	-11.02	7.44	18.46	-6.72	13.38	20.10	-6.05	13.75	19.80
l 12 sec	-10.62	21.48	32.16	-11.54	7.78	19.32	-2.89	30.30	33.19	-6.33	14.39	20.72
10 sec	-13.10	4.52	17.62	-14.15	9.54	23.69	-10.96	7.92	11.88	-7.76	17.66	25.47
m 26 sec	-10.07	-2.05	9.07	-10.87	7.33	18.20	-10.31	-1.33	8.98	-5.97	13.56	19.53
140 sec	-6.84	8.67	15.51	-7.38	4.98	12.36	-3.56	12.40	16.16	-4.05	9.21	13.26
n 33 sec	-1.02	37.44	38.84	-1.54	1.04	2.58	11.36	50.36	39.00	-0.84	1.92	7.76
54 sec	-3.32	64.01	67.33	-3.59	2.42	6.01	18.59	86.24	67.65	-1.97	4.48	6.45

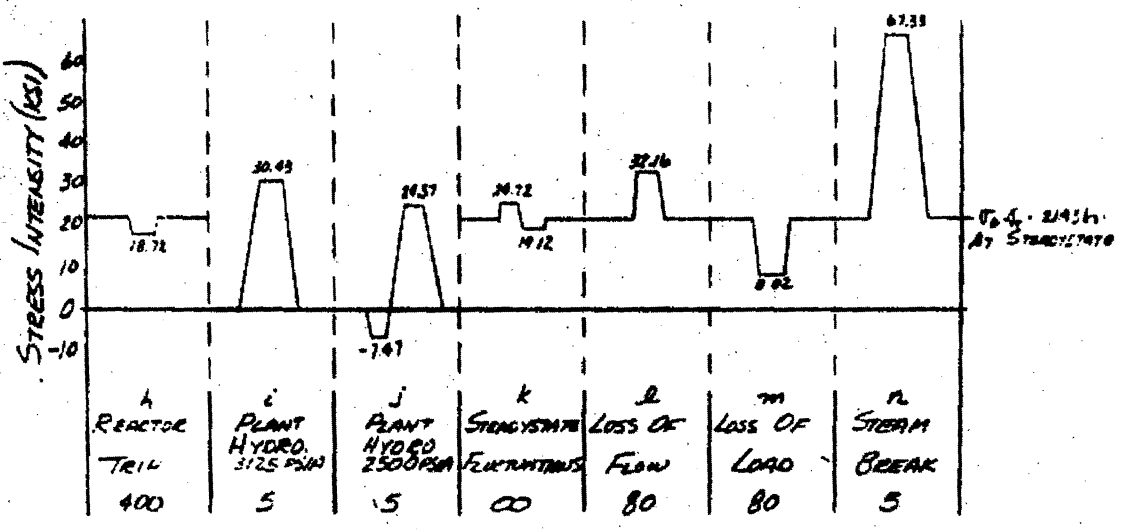
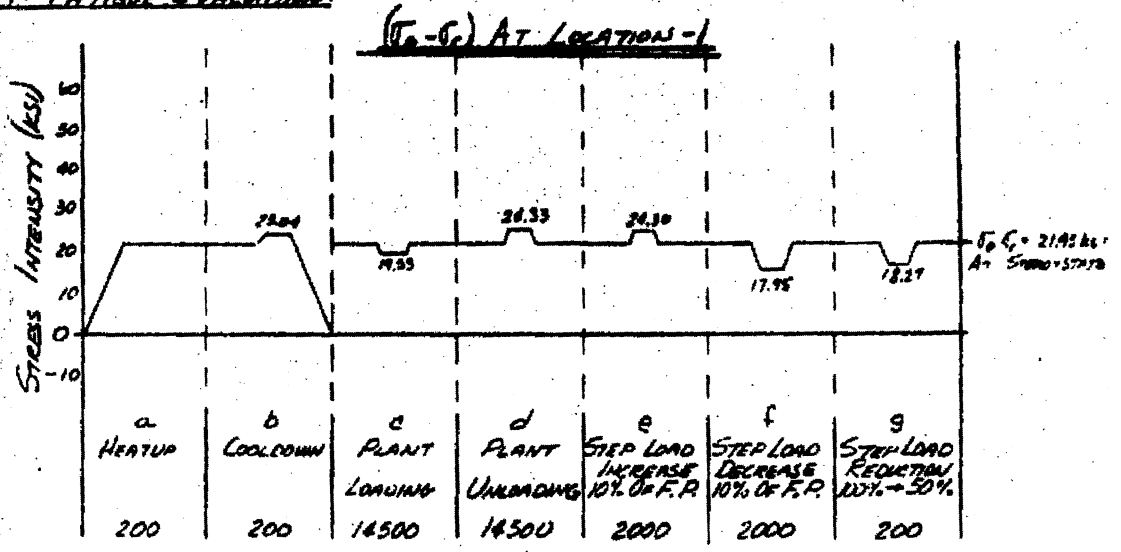
COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-P | A 209
SHEET 21 OF 28

CHARGE NO. _____ DATE MAY 26, 1966 BY COYRELL
DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL WALL TRANSITION CHECK DATE MAY 26, 1966 BY ALEXANDER

S- DETAILED ANALYSIS:

F- FATIGUE EVALUATION:



SWAY	SWAY	NUMBER OF OCCURRENCES	S _{ALT}	N ⁰	U
67.33	-7.47	5	37.4	1000	0.00050
32.16	0	80	16.1	250,000	1.00000
30.43	0	5	15.2	30000	0.00001
24.72	0	115	12.4	∞	0

* FROM FIG. N-45(A)
REFERENCE I

U_{NORMAL} = 0.00106

COMBUSTION ENGINEERING, INC.

ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

CHARGE NO. _____

NUMBER S-202-P | A 290

SHEET 22 OF 24

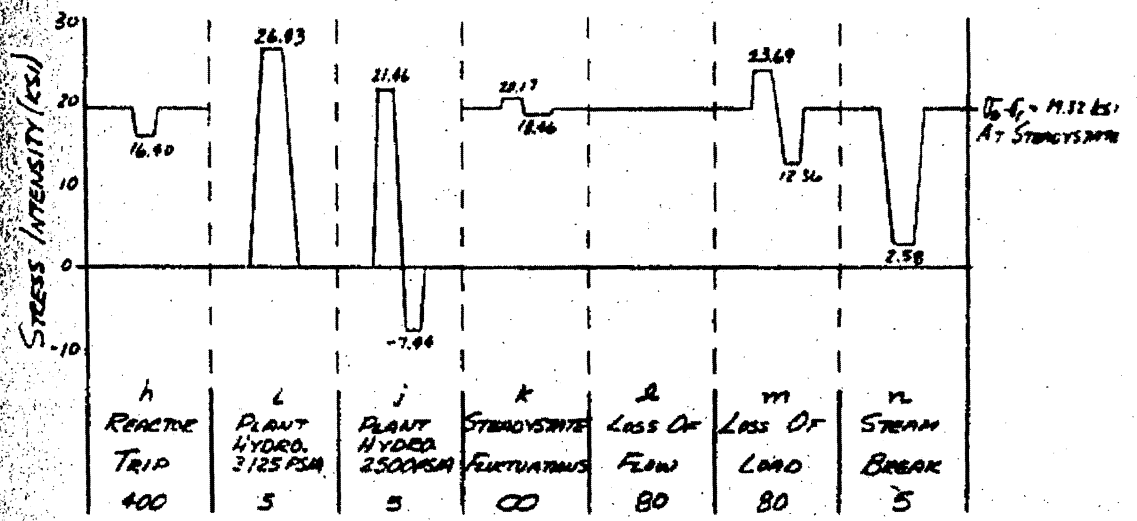
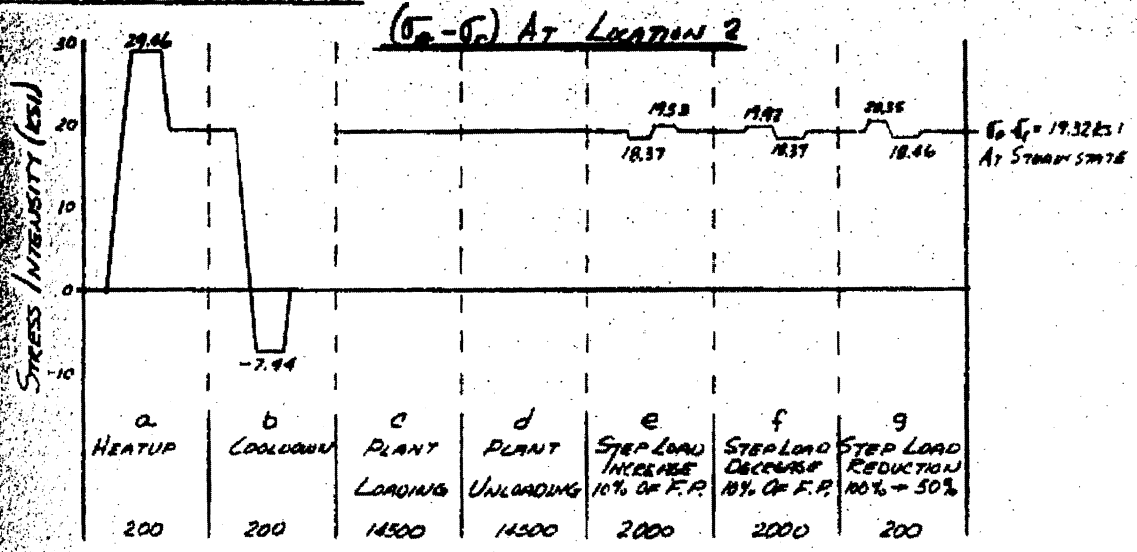
DATE MAY 26, 1966 BY LORENCE

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF
THE VESSEL WALL TRANSITION

CHECK DATE MAY 26, 1966 BY ALEXANDER

S. DETAILED ANALYSIS:

F. FATIGUE EVALUATION:



S_{MAX}	S_{MIN}	NUMBER OF CYCLES	S_{MEAN}	N^*	U
29.46	-7.44	200	18.5	185,000	0.00148
26.83	-7.44	5	17.1	190,000	0.00003
23.69	2.58	5	10.6	∞	0

* FROM FIG. N-415(A)
REFERENCE 1
 $U_{NORMAL} = 0.00151$

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-P | A291
SHEET 25 OF 26

CHARGE NO. _____

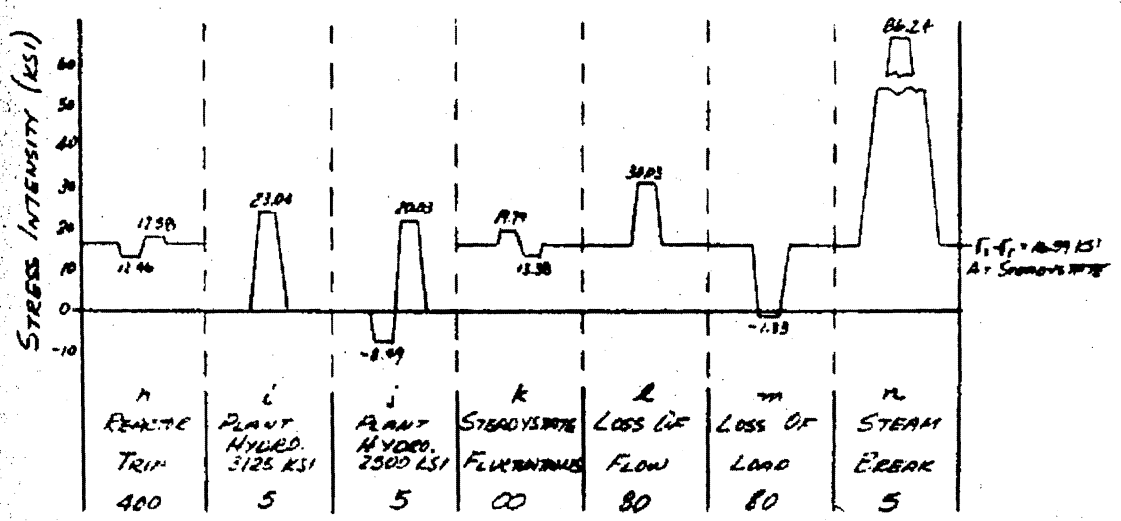
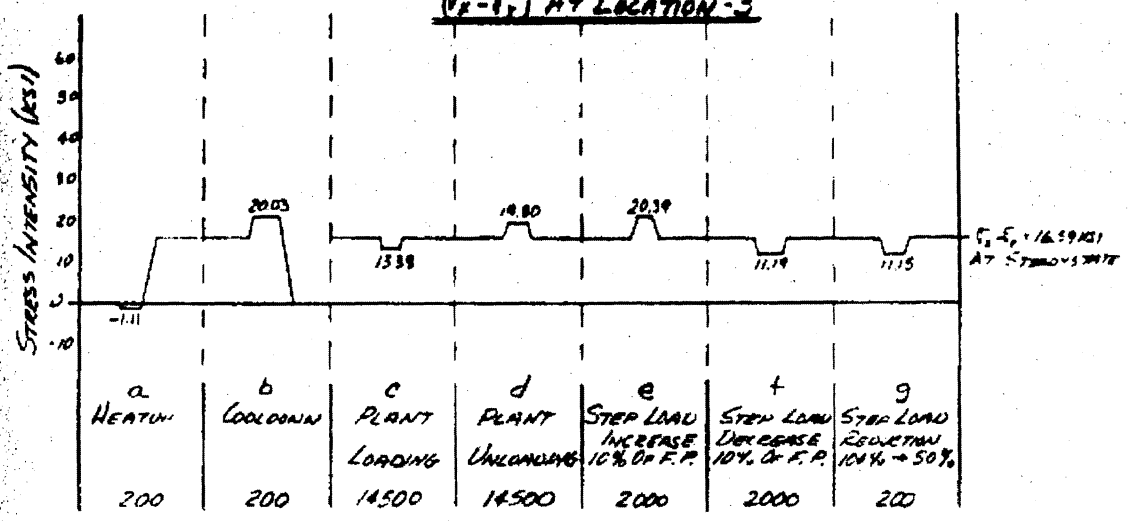
DATE Mar 26, 1966 BY COOPER

DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL WALL TRANSITION

CHECK DATE 1/11/26/66 BY ALWOODER

5. DETAILED ANALYSIS:
F. FATIGUE EVALUATION:

($\sigma_1 - \sigma_2$) AT LOCATION -3



Sum.	S _{MIN}	NUMBER OF INCREASES	S _{AUT}	N ^a	U
16.24	2.49	5	47.4	500	3.00100
30.03	-1.33	80	15.7	29000	0.00027
23.24	-1.11	5	121	∞	0

* FROM FIG. N-815(A)
REFERENCE 1
U_{INTERNAL} = 0.00127

COMBUSTION ENGINEERING, INC.

ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-202-P | A 242

SHEET 24 OF 24

CHARGE NO. _____

DATE MAY 26, 1966 BY CK/KMPL

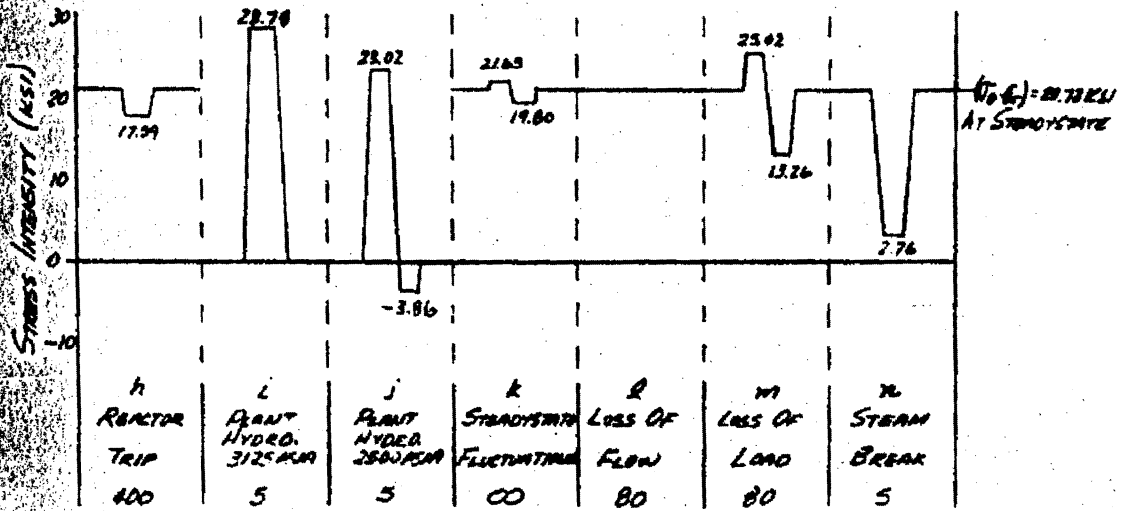
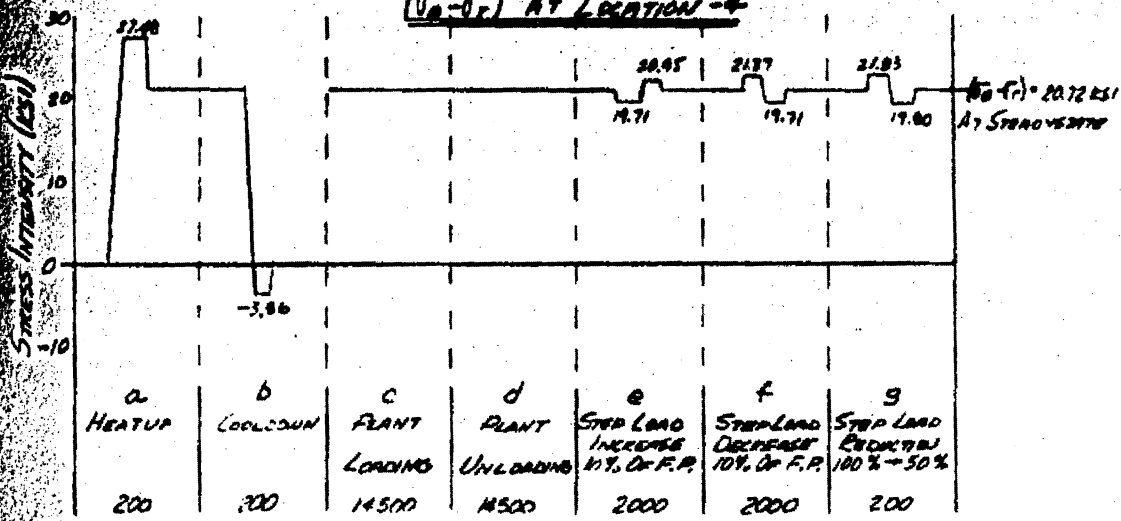
DESCRIPTION STRUCTURAL AND FATIGUE ANALYSIS OF THE VESSEL W/OUT TRANSITION

CHECK DATE MAY 26, 1966 BY ALEXANDER

5- DETAILED ANALYSIS:

F. FATIGUE EVALUATION:

($\sigma_a - \sigma_r$) AT LOCATION -4



S_{max}	S_{min}	NUMBER OF OCCURRENCES	S_{avg}	N^*	U
27.18	-3.86	5	16.3	230000	0.00002
27.18	-3.86	200	15.7	240000	0.00064
25.42	2.76	5	11.3	∞	0

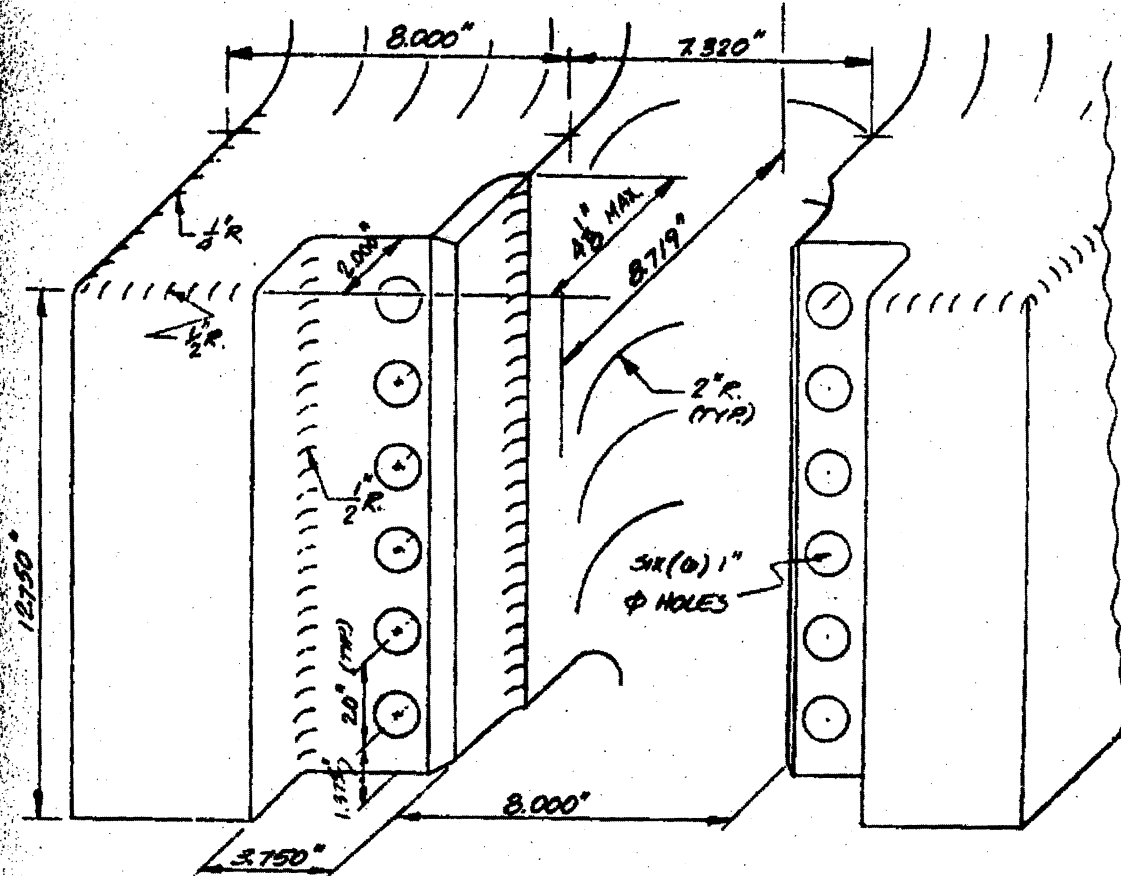
* FROM FIG. N-415(A) REFERENCE 1...
 $U_{total} = 0.00071$

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-201-P / A-29
SHEET 5 OF 91
DATE 1-27-67 BY ALEXANDER
CHECK DATE 1-27-67 BY COMBLE

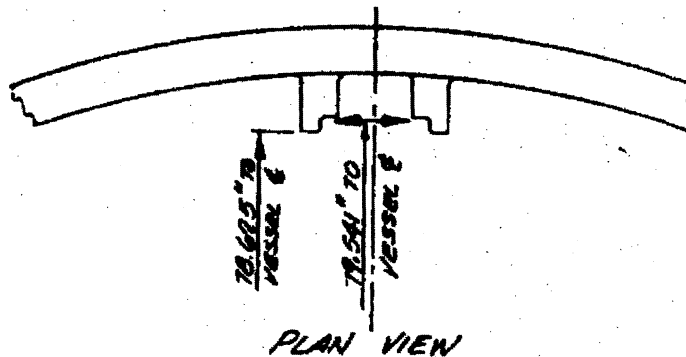
CHARGE NO. _____
DESCRIPTION STRUCTURAL ANALYSIS OF
CORE SUPPORT BASE

5. DETAILED ANALYSIS
2. SYSTEM GEOMETRY



SEE C.E.DWG E-337-050

MATERIAL:
INCONEL



1. SHOW THAT THE STRESS INTENSITY DERIVED FROM PRIMARY MEMBRANE (GENERAL OR LOCAL) PLUS PRIMARY BENDING STRESSES PRODUCED BY DESIGN PRESSURE AND OTHER MECHANICAL LOADS IS LESS THAN $1.5 S_m$.
2. SHOW THAT THE RANGE OF STRESS INTENSITY AT EACH POINT DUE TO THE COMBINATION OF MECHANICAL LOADS PLUS THERMAL EFFECTS (NEGLECTING STRESS CONCENTRATIONS) IS LESS THAN $3S_m$.
3. SHOW THAT EACH POINT MEETS THE REQUIREMENTS FOR PEAK STRESS INTENSITIES GIVEN IN N-415.5 OF THE A.S.M.E. CODE. THE PROCEDURE WILL BE THAT DESCRIBED IN N-415.2 AND N-416.2 OF SECTION III.

(c) SYSTEM ALLOWABLES

1. A MOMENTARY VERTICAL LOAD OF 250 KIPS DURING INSERTION OF THE CORE.
2. A NON-CYCLIC STEADY LOADING OF 125 KIPS IN THE CIRCUMFERENTIAL DIRECTION.
3. A VERTICAL THERMAL GROWTH FRICTION LOAD OF 125 KIPS BETWEEN THE KEY ON THE CORE BARREL AND THE PAD AT ONE VERTICAL INTERFACE ONLY.
4. A CYCLIC LOADING OF PLUS/MINUS 100 KIPS ACTING IN THE CIRCUMFERENTIAL DIRECTION FOR AN INFINITE NUMBER OF CYCLES.
5. AN ADDITIONAL CYCLE LOAD OF 125 KIPS IN THE CIRCUMFERENTIAL DIRECTION DUE TO BENTHAQUE CONSIDERATIONS.
6. THE OPERATING TRANSIENTS DESCRIBED IN E-502C # 676208.

THE SYSTEM SHOWN ON SHEET 5 WILL BE ANALYZED FOR THE FOLLOWING LOADINGS:

(b) SYSTEM LOADS

5. DETAILED ANALYSIS

CORE SUPPORT PADS

COMBUSTION ENGINEERING, INC.
 ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.
 CHANGE NO. _____
 STRUCTURAL ANALYSIS OF _____
 CORE SUPPORT PADS
 NUMBER 5-81-P-1A94
 SHEET 6 OF 31
 DATE 1-27-67 BY ALAN COOPER
 CHECK DATE 1-27-67 BY CARL

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

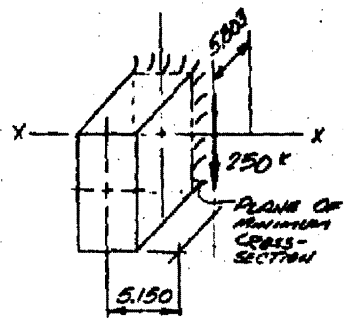
NUMBER 5-201-P | 1A295
SHEET 7 OF 31
DATE 1-27-67 BY ALEXANDER
CHECK DATE 1-27-67 BY CAROLIE

CHARGE NO. _____
DESCRIPTION STRUCTURAL ANALYSIS OF
COPE SUPPORT PARS

5. DETAILED ANALYSIS

d) STRESS DURING INSERTION OF COPE

CONSIDER LOADING 1 SHOWN ON SHEET 6. DURING INSTALLATION OF THE COPE, IF ALIGNMENT FAILS TO BE SMOOTH, A VERTICAL LOAD OF 250 KIPS OF MOMENTARY DURATION WILL BE TRANSMITTED TO ONE LUG AS SHOWN BELOW. STRESSES WERE CONSIDERED AT THE JUNCTURE OF THE LUG TO THE VESSEL WALL AND AT THE PLANE OF MINIMUM CROSS-SECTION. IT WAS DETERMINED THAT THE CONTROLLING LOCATION WAS THE PLANE OF MINIMUM CROSS-SECTION.



FOR THE MINIMUM CROSS-SECTION:

$$I_x = \frac{bh^3}{12} = \frac{8(12.75)^3}{12} = 1381.781 \text{ in.}^4$$

AND, FROM PG. 289, REF. 8 -

$$\frac{b}{c} = \frac{12.75}{8.0} = 1.59; \alpha = 0.234$$

BENDING STRESS

$$\sigma = \frac{Mc}{I_x}$$

SHORT SIDE

$$\frac{5208(250)6.375}{1381.781} = 6.7 \text{ KSI}$$

LONG SIDE

0

TORSION

$$\tau_s = \frac{T}{\alpha bc^2}$$

$$\frac{250(5.15)}{.234(8)(12.75)^2} = 4.2 \text{ KSI}$$

$$\frac{250(5.15)}{.234(12.75)^2} = 6.7 \text{ KSI}$$

DIRECT SHEAR

$$\tau_s = 15 \frac{V}{A}$$

0

$$15 \frac{250}{102.0} = 37 \text{ KSI}$$

CONSIDER MID-POINT OF SHORT SIDE
 $\sigma_1 = \frac{\sigma}{2} + \sqrt{\frac{\sigma^2}{4} + \tau^2} = 8.7 \text{ KSI}$
 $\sigma_2 = \frac{\sigma}{2} - \sqrt{\frac{\sigma^2}{4} + \tau^2} = -2.0 \text{ KSI}$
 $\tau_1 - \sigma_2 = 8.7 - (-2.0) = 10.7 \text{ KSI} \times 1.15 = 12.3 \text{ KSI}$

CONSIDER MID-POINT OF LONG SIDE
 $\sigma = 0$
 $\tau = 6.7 + 3.7 = 10.4 \text{ KSI} < 0.85 \times 12.3 = 10.6 \text{ KSI}$

COMBUSTION ENGINEERING, INC.

ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER S-201-P | A24

SHEET 8 OF 31

CHARGE NO. _____

DATE 1-27-67 BY ALEXANDER

DESCRIPTION STRUCTURAL ANALYSIS OF

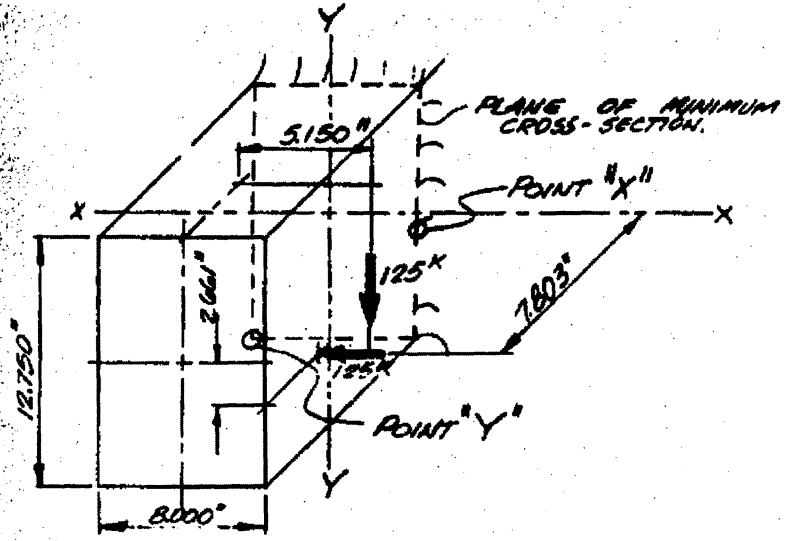
CHECK DATE 1-27-67 BY COOPER

CORE SUPPORT PADS

5. DETAILED ANALYSIS

(c) STRESS DUE TO STEADY SIDE LOAD OF 125 KIPS,
STEADY VERTICAL LOAD (THERMAL GROWTH) OF 125 KIPS
AND DESIGN PRESSURE

AT THE MINIMUM CROSS-SECTION



CONSIDERATION OF POINTS ALONG THE PERIPHERY OF THE PLANE OF MINIMUM CROSS-SECTION SHOWED THAT POINT "X" AS SHOWN ABOVE WAS THE CONTROLLING LOCATION FOR THIS PLANE. THE FOLLOWING IS A TABULATION OF THE STRESSES AT POINT "X".

$$T = 125(5.150) + 125(2.661) = 976.375 \text{ IN-KIPS}$$

$$\frac{b}{c} = \frac{12.750}{8.0} = 1.59, \quad \alpha = 0.834 \text{ (PAGE 289, REF B)}$$

$$I_{yy} = \frac{bh^3}{12} = 544 \text{ IN}^4$$

$$\text{BENDING STRESS} = \frac{Mc}{I_y} = + \frac{125(5.800)(4.0)}{544} = +5.3 \text{ KSI}$$

$$\text{TORSIONAL SHEAR} = \frac{T}{\alpha bc^2} = \frac{976.375}{.834(12.75)(8.0)^2} = 5.1 \text{ KSI}$$

$$\text{DIRECT SHEAR} = 1.5 \frac{V}{A} = 1.5 \frac{125}{102} = 1.8 \text{ KSI}$$

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-201-P | A 197
SHEET 9 OF 31
DATE 1-27-67 BY ALEXANDER
CHECK DATE 1-27-67 BY CONRUB

CHARGE NO. _____
DESCRIPTION STRUCTURAL ANALYSIS OF COKE
SUPPORT PADS

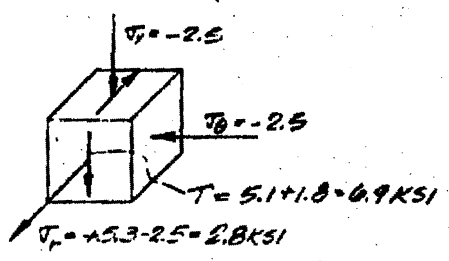
5. DETAILED ANALYSIS

c) STRESS DUE TO STEADY SIDE LOAD OF 125 KIIPS,
STEADY VERTICAL LOAD (THERMAL GROWTH) OF 125 KIIPS
AND DESIGN PRESSURE

THE DESIGN PRESSURE IS 2.5 KSI WHICH GIVES -

$$\sigma_i = \sigma_o = \tau_r = -2.5 \text{ KSI}$$

CONSIDER A STRESS BLOCK AT POINT "X".



$$\sigma_1 = \frac{\sigma_r + \sigma_x}{2} + \sqrt{\left(\frac{\sigma_r - \sigma_x}{2}\right)^2 + \tau^2} = 7.5 \text{ KSI}$$

$$\sigma_2 = \frac{\sigma_r + \sigma_x}{2} - \sqrt{\left(\frac{\sigma_r - \sigma_x}{2}\right)^2 + \tau^2} = -7.2 \text{ KSI}$$

$$\sigma_3 = -2.5 \text{ KSI}$$

$$S.I._{MAX} = (\sigma_1 - \sigma_2) = 7.5 - (-7.2) = 14.7 \text{ KSI} < 1.5 S_m = 35.0$$

CRITERION S.C.1

AT THE VESSEL WALL -

CONSIDERATION OF POINTS AT THE JUNCTURE OF THE PAD TO THE VESSEL WALL SHOWS THAT THE CONTROLLING LOCATION IS POINT "Y" (SEE SHEET B).

$$I_{y-y} = \frac{bh^3}{12} \quad \text{WHERE ; } b = 12.750 + 2(.75)2 = 15.75$$

$$h = 8.000 + 2(.75)2 = 11.0$$

$$= \frac{15.75(11)^3}{12} = 1746.938 \text{ IN.}^4$$

$$I_{x-x} = \frac{11.0(15.75)^3}{12} = 3581.402 \text{ IN.}^4$$

$$T = 976375 \text{ IN.-KIIPS}$$

$$\text{BENDING STRESS} = \frac{M_c}{I_{x-x}} = \frac{M_c}{I_{y-y}} = \frac{125(7.000)2075}{3581.402} = \frac{125(7.000)55}{1746.938} = 5.2 \text{ KSI}$$

TORSION AND DIRECT SHEAR = 0

COMBUSTION ENGINEERING, INC.

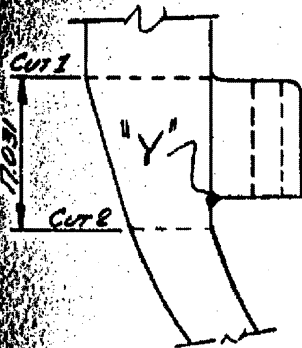
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

CHARGE NO. _____

NUMBER S-201-P/A 199SHEET 12 OF 31DATE 1-27-67 BY ALEXANDERDESCRIPTION STRUCTURAL ANALYSIS OFCHECK DATE 1-23-67 BY CAUDLECORE SUPPORT PADSB. DETAILED ANALYSIS

C) STRESS DUE TO STEADY SIDE LOAD OF 125 KIPS,
STEADY VERTICAL LOAD (THERMAL GROWTH) OF 125 KIPS
AND DESIGN PRESSURE

EXPRESSIONS FOR THE STRESSES AT THE INSIDE SURFACE OF THE VESSEL WERE PREVIOUSLY DERIVED AT CUTS 1 AND 2 (SEE BELOW) IN C.E. CALCULATION S-200-P. PRESSURE STRESSES AT POINT "Y" WILL BE DETERMINED BY PERFORMING A LINEAR INTERPOLATION BETWEEN CUTS 1 AND 2.



CUT	σ_x	σ_θ	σ_r
1	5.285P	8.671P	-P
2	9.910P	8.465P	-P

} FROM SHEET-14
OF S-200-P

INTERPOLATION YIELDS THE FOLLOWING STRESSES FOR POINT "Y":

$$\begin{aligned} \sigma_x &= 9.358P \quad \text{FOR } P=2.5; & \sigma_x &= 23.4 \text{ KSI} \\ \sigma_\theta &= 8.490P & \sigma_\theta &= 21.2 \text{ KSI} \\ \sigma_r &= -P & \sigma_r &= -2.5 \text{ KSI} \end{aligned}$$

COMBINING PRESSURE AND BENDING STRESSES -

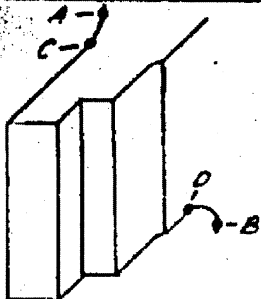
$$\sigma_x = 23.4 \text{ KSI}$$

$$\sigma_\theta = 21.2 \text{ KSI}$$

$$\sigma_r = -5.2 - 2.5 \text{ KSI} = -7.7 \text{ KSI}$$

$$S.I. = (\sigma_\theta - \sigma_r) = \frac{31.1 \text{ KSI}}{1.5} < 1.5 \sigma_u = 350 \text{ KSI}$$

CRITERION S.C.1

F) STRESSES1. UNCONCENTRATED

THE LOCATIONS SHOWN AT LEFT WILL BE INVESTIGATED FOR THE FOLLOWING LOADS:

- 1) A STEADY SIDE LOAD OF 125 K
- 2) A CYCLIC LOADING OF 2KIP IN THE CIRCUMFERENTIAL DIRECTION.
- 3) A CYCLIC LOADING OF 125K IN THE CIRCUMFERENTIAL DIRECTION DUE TO EARTHQUAKE CONSIDERATIONS
- 4) THE EFFECTS OF THE OPERATING TRANSIENTS DESCRIBED IN E-SPEC. REF. 19

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-201-P | AB99
SHEET 11 OF 31
DATE 1-27-67 BY ALEXANDER
CHECK DATE 1-27-67 BY CAULDE

CHARGE NO. _____
DESCRIPTION STRUCTURAL ANALYSIS OF
COFE SUPPORT PADS

5. DETAILED ANALYSIS

f.) STRESSES

1. UNCONCENTRATED

THERMAL STRESSES - *
POINTS "A" & "B" -

$$\sigma_x = \sigma_\theta = \frac{E\alpha(T_m - T)}{1 - \nu}$$

POINTS "C" & "D" -

$$\sigma_x = \sigma_r = \frac{E\alpha(T_m - T)}{1 - \nu}$$

WHERE; $E\alpha$ = YOUNG'S MODULUS X COEFFICIENT OF THERMAL EXPANSION, TAKEN AS 0.214 FOR POINTS "A" & "B" AND 0.254 FOR POINTS "C" & "D".

T_m = MEAN TEMPERATURE AT STEADY STATE.

T = INSTANTANEOUS REACTOR INLET COOLANT TEMPERATURE DURING TRANSIENT.

ν = POISSON'S RATIO, TAKEN AS 0.3.

PRESSURE STRESSES -

POINT "A" -

$$\sigma_x = 5.285 P \text{ FROM PG. 10}$$

$$\sigma_\theta = 8.671 P$$

$$\sigma_r = -P$$

POINT "B" -

$$\sigma_x = 9.358 P \text{ FROM PG. 10}$$

$$\sigma_\theta = 8.490 P$$

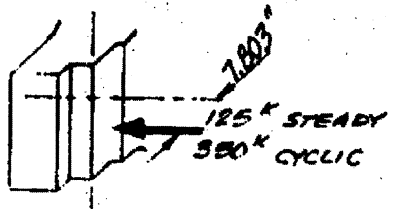
$$\sigma_r = -P$$

POINTS "C" & "D" -

$$\sigma_x = \sigma_\theta, \sigma_r = -P$$

STRESSES DUE TO APPLIED LOADS -

COMBINING THE CYCLIC AND STEADY LOADS AS GIVEN ON PAGE 10 YIELDS THE FOLLOWING APPLIED LOADING:



* NOTE - ALL THERMAL STRESSES CONSERVATIVELY TREATED AS "SKIN" TYPE STRESSES.

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

NUMBER 5-201-P | A300

SHEET 12 OF 31

CHARGE NO. _____

DATE 1-27-67 BY ALEXANDER

DESCRIPTION STRUCTURAL ANALYSIS OF
CORE SUPPORT PADS

CHECK DATE 1-27-67 BY CAUDLE

5 DETAILED ANALYSIS

f.) STRESSES

1. UNCONCENTRATED

STRESSES DUE TO APPLIED LOADS -

POINT "A" - (VESSEL WALL)

$$\sigma_r = - \frac{Mc}{I_{yy}} = - \frac{(390)(125)7.803(5.5)}{1746.938} = \begin{matrix} -8.6 \text{ KSI} \\ -9.1 \text{ KSI} \end{matrix}$$

POINT "B" - (VESSEL WALL)

$$\sigma_r = \begin{matrix} +8.6 \text{ KSI} \\ +3.1 \text{ KSI} \end{matrix}$$

POINT "C" -

$$\sigma_r = - \frac{Mc}{I_{yy}} = - \frac{(390)(125)5.803(4)}{544} = \begin{matrix} -14.9 \text{ KSI} \\ -5.3 \text{ KSI} \end{matrix}$$

POINT "D" -

$$\sigma_r = \begin{matrix} +14.9 \text{ KSI} \\ +5.3 \text{ KSI} \end{matrix}$$

SHEAR STRESSES NEGLECTED.

NOTE: THE FOLLOWING TABLES SUMMARIZE THE STRESSES AND STRESS INTENSITIES FOR THE FOUR LOCATIONS AS SHOWN ON SHEET 10. COMPARISON WITH 35_m ALLOWABLES (CRITERION B-C-2) IS GIVEN AT THE END OF EACH TABLE. FOR THE 35_m CONSIDERATION, THE FOLLOWING TRANSIENTS ARE NOT CONSIDERED:

- 1- "LOSS OF FLOW"
- 2- "STREAM BREAK"
- 3- "LOSS OF LOAD"

SUMMARY OF STRESSES AND STRESS INTENSITIES AT LOCATION A

TRANSIENT	TIME	PRESS (KSI)	$(\sigma_a - \tau)$	WELDED	PRESSURE STRESSES			LOAD	TOTAL STRESSES			STRESS INTENSITIES		
				STRESS	σ_x	σ_y	σ_z	STRESS	σ_x	σ_y	σ_z	$\sigma_x - \sigma_y$	$\sigma_x - \sigma_z$	$\sigma_y - \sigma_z$
STEADY STATE	—	2.25	0	0	11.9	19.5	-2.3	-3.1	11.9	19.5	-5.4	-7.6	17.3	24.9
HEAT-UP	4.47 HRS.	2.25	-43	-13.1	11.9	19.5	-2.3	-3.6	-1.2	6.4	-10.9	-7.6	9.7	17.3
		2.25	-43	-13.1	11.9	19.5	-2.3	-3.1	-1.2	6.4	-5.4	-7.6	4.2	11.8
COOLDOWN	4.47 HRS.	0.32	43	13.1	1.7	2.8	-0.3	-8.6	14.8	15.9	-8.9	-1.1	23.7	24.8
		0.32	43	13.1	1.7	2.8	-0.3	-3.1	14.8	15.9	-3.4	-1.1	18.2	19.3
PLANT LOADING	20 MIN	2.25	-7.8	-2.4	11.9	19.5	-2.3	-8.6	9.5	17.1	-10.9	-7.6	20.4	28.0
		2.25	-7.8	-2.4	11.9	19.5	-2.3	-3.1	9.5	17.1	-5.4	-7.6	14.9	22.5
PLANT UNLOADING	20 MIN	2.25	7.8	2.4	11.9	19.5	-2.3	-8.6	14.3	21.9	-10.9	-7.6	25.2	32.8
		2.25	7.8	2.4	11.9	19.5	-2.3	-3.1	14.3	21.9	-5.4	-7.6	19.7	27.3
STEP LOAD INCREASE	100 SEC	2.14	11.2	3.4	11.3	18.6	-2.1	-8.6	14.7	22.0	-10.7	-7.3	25.4	32.7
10% OF FULL POWER	220 SEC	2.28	1.7	0.5	12.0	19.8	-2.3	-8.6	12.5	20.3	-10.9	-7.8	23.4	31.2
STEP LOAD DECREASE	100 SEC	2.32	-9.3	-2.8	12.3	20.1	-2.3	-8.6	9.5	17.3	-10.9	-7.8	20.4	28.2
10% OF FULL POWER	240 SEC	2.24	-13.3	-4.1	11.9	19.6	-2.3	-3.1	7.8	15.5	-10.9	-7.7	18.7	26.4
LOSS OF FLOW	12 SEC	2.25	33.3	10.2	11.9	19.5	-2.3	-8.6	22.1	29.7	-10.9	-7.6	33.0	40.6
STREAM BREAK	54 SEC	0.70	197	60.2	3.7	6.1	-0.7	-8.6	63.9	66.3	-9.3	-2.4	73.2	79.6

DETAILED ANALYSIS
 OF STRESSES - APPROXIMATED

COMBUSTION ENGINEERING, INC.
 ENGINEERING DEPARTMENT, QUALIFICATION TEST
 CHANGE NO. _____
 STRUCTURAL ANALYSIS OF CORE SUPPORT PINS
 NUMBER 5-82-1430
 SHEET 12 OF 21
 DATE 1-21-62 BY [Signature]
 CHECK DATE 1-27-62 BY [Signature]

COMBUSTION ENGINEERING, INC.
ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.

CHARGE NO. _____
DESCRIPTION: STRUCTURAL ANALYSIS OF CORE SUPPORT RIBS
DATE: 1-27-67 BY: ALEXANDER
CHECK DATE: 1-27-67 BY: CAWNE

5. DETAILED ANALYSIS
STRESSES - LINEAR/INTERPOLATED

TRANSIENT	TIME	PRESS. (KSI)	$(T_m - T)$	TEMP. STRESS σ_{T-0}	PRESSURE STRESSES			LOAD STRESS σ_T	TOTAL STRESSES			STRESS INTENSITIES		
					σ_c	σ_b	σ_r		σ_c	σ_b	σ_r	σ_{c-b}	σ_{r-b}	σ_{c-r}
STEP LOAD	2 MIN.	2.57	-12.0	-3.7	12.5	20.6	-2.4	-8.6	8.8	16.9	-11.0	-8.1	19.8	27.9
REDUCTION	2 MIN.	2.37	-12.0	-3.7	12.5	20.6	-2.4	-3.1	8.8	16.9	-5.5	-8.1	14.3	22.4
FROM 100%	32 MIN.	2.35	-15.0	-4.6	12.4	20.4	-2.4	-8.6	7.8	15.8	-11.0	-8.0	18.8	26.8
TO 50%	32 MIN.	2.35	-15.0	-4.6	12.4	20.4	-2.4	-3.1	7.8	15.8	-5.5	-8.0	13.3	21.3
FULL POWER	N/A MIN.	2.15	0	0	11.4	18.6	-2.2	-8.6	11.4	18.6	-10.8	-7.2	22.2	29.4
	N/A MIN.	2.15	0	0	11.4	18.6	-2.2	-3.1	11.4	18.6	-5.3	-7.2	16.7	23.9
REACTOR	10 SEC.	2.22	9.5	-2.9	11.7	19.2	-2.2	-8.6	8.8	16.3	-10.8	-7.5	19.6	27.1
TRIP FROM	10 SEC.	2.22	9.5	-2.9	11.7	19.2	-2.2	-3.1	8.8	16.3	-5.3	-7.5	14.1	21.6
FULL	65 SEC.	1.91	8.5	2.6	10.1	16.6	-1.9	-8.6	12.7	19.2	-10.5	-6.5	23.2	29.7
POWER	65 SEC.	1.91	8.5	2.6	10.1	16.6	-1.9	-3.1	12.7	19.2	-5.0	-6.5	17.7	24.2
PLANT HYDRO	220 MIN.	3.13	0	0	16.5	27.1	-3.1	-8.6	16.5	27.1	-11.7	-10.6	28.2	38.8
AT 2500 PSIA	220 MIN.	3.13	0	0	16.5	27.1	-3.1	-3.1	16.5	27.1	-6.2	-10.6	22.7	33.3
PLANT HYDRO	~	2.50	-43	-13.1	13.2	21.7	-2.5	-8.6	0.1	8.6	-11.1	-8.5	11.2	19.7
AT	~	2.50	-43	-13.1	13.2	21.7	-2.5	-3.1	0.1	8.6	-5.6	-8.5	5.7	14.2
AT 2500 PSIA	~	0.32	43	13.1	1.7	2.8	-0.3	-8.6	14.8	15.9	-8.9	-1.1	23.7	24.8
	~	0.32	43	13.1	1.7	2.8	-0.3	-3.1	14.8	15.9	-3.4	-1.1	18.2	19.3
STEADY STATE	~	2.35	6.0	1.8	12.4	20.4	-2.4	-8.6	14.2	22.2	-11.0	-8.0	25.2	33.2
FLUCTUATIONS	~	2.35	6.0	1.8	12.4	20.4	-2.4	-3.1	14.2	22.2	-5.5	-8.0	19.7	27.7
OF PRESS.	~	2.15	-6.0	-1.8	11.4	18.6	-2.2	-8.6	9.6	16.8	-10.8	-7.2	20.4	27.6
AND TEMP.	~	2.15	-6.0	-1.8	11.4	18.6	-2.2	-3.1	9.6	16.8	-5.3	-7.2	14.9	22.1
LOSS	10 SEC.	2.76	-30.2	-9.2	14.6	23.9	-2.8	-8.6	5.4	14.7	-11.4	-9.3	16.8	26.1
OF	10 SEC.	2.76	-30.2	-9.2	14.6	23.9	-2.8	-3.1	5.4	14.7	-5.9	-9.3	11.3	20.6
LOAD	28 SEC.	2.12	-41.2	-12.6	11.2	18.4	-2.1	-8.6	-1.4	5.8	-10.7	-7.2	9.3	16.5
	28 SEC.	2.12	-41.2	-12.6	11.2	18.4	-2.1	-3.1	-1.4	5.8	-5.2	-7.2	3.8	11.0
	140 SEC.	1.44	4.8	1.5	7.6	12.5	-1.4	-8.6	9.1	14.0	-10.0	-4.9	19.1	24.0
	140 SEC.	1.44	4.8	1.5	7.6	12.5	-1.4	-3.1	9.1	14.0	-4.5	-4.9	13.6	18.5

SI Unit Range - $\sigma_{c-b} = 30.8 \text{ KSI} < 35 \text{ KSI} = 80 \text{ KSI}$ (Reference S-2-2)

SUMMARY OF STRESSES AND STRESS INTENSITIES AT LOCATION B

TRANSIENT	TIME	PRESS (KSI)	(h-7)	THERM STRESS		PRESSURE STRESSES			LOAD STRESS	TOTAL STRESSES			STRESS INTENSITIES		
				σ_1	σ_2	σ_3	σ_4	σ_5	σ_6	σ_7	σ_8	σ_9	σ_{10}	σ_{11}	σ_{12}
STEADY STATE	~	2.25	0	0	21.1	19.1	-2.3	3.1	21.1	19.1	0.8	2.0	20.3	18.3	
HEAT-UP	4.47 HRS.	2.25	-43	-13.1	21.1	19.1	-2.3	8.6	8.0	6.0	6.3	2.0	1.7	-0.3	
		2.25	-43	-13.1	21.1	19.1	-2.3	3.1	8.0	6.0	0.8	2.0	7.2	5.2	
COOLDOWN	4.47 HRS.	0.32	43	13.1	3.0	2.7	-0.3	8.6	16.1	15.8	8.3	0.3	7.8	7.5	
		0.32	43	13.1	3.0	2.7	-0.3	3.1	16.1	15.8	2.8	0.3	13.3	13.0	
PLANT LOADING	20 MIN	2.25	-7.8	-2.4	21.1	19.1	-2.3	8.6	18.7	16.7	6.3	2.0	12.4	10.4	
		2.25	-7.8	-2.4	21.1	19.1	-2.3	3.1	18.7	16.7	0.8	2.0	17.9	15.9	
PLANT UNLOADING	20 MIN	2.25	7.8	2.4	21.1	19.1	-2.3	8.6	23.5	21.5	6.3	2.0	17.2	15.2	
		2.25	7.8	2.4	21.1	19.1	-2.3	3.1	23.5	21.5	0.8	2.0	22.7	20.7	
STEP LOAD INCREASE	100 SEC.	2.14	11.2	3.4	20.0	18.2	-2.1	8.6	23.4	21.6	6.5	1.8	16.9	15.1	
10% OF FULL POWER	220 SEC.	2.28	1.7	0.5	21.3	19.4	-2.3	8.6	21.8	19.9	6.3	1.9	15.5	13.6	
STEP LOAD DECREASE	100 SEC.	2.32	-9.3	-2.8	21.7	19.7	-2.3	8.6	18.9	16.9	6.3	2.0	12.6	10.6	
10% OF FULL POWER	240 SEC.	2.24	-13.3	-4.1	21.1	19.2	-2.3	3.1	17.0	15.1	0.3	1.9	10.7	8.8	
STEP LOAD INCREASE	100 SEC.	2.14	-1.3	-0.4	20.0	18.2	-2.1	8.6	19.6	17.8	6.5	1.8	13.1	11.3	
10% OF FULL POWER	260 SEC.	2.14	-1.3	-0.4	20.0	18.2	-2.1	3.1	19.6	17.8	1.0	1.8	18.6	16.8	
LOSS OF FLOW	12 SEC.	2.25	33.3	10.2	21.1	19.1	-2.3	8.6	31.3	29.3	6.3	2.0	25.0	23.0	
		2.25	33.3	10.2	21.1	19.1	-2.3	3.1	31.3	29.3	0.8	2.0	30.5	28.5	
STEAM BREAK	54 SEC.	0.70	197	60.2	6.6	5.9	-0.7	8.6	66.8	66.1	7.9	0.7	58.9	58.2	
		0.70	197	60.2	6.6	5.9	-0.7	3.1	66.8	66.1	2.4	0.7	64.4	63.7	

5. DETAILED ANALYSIS
 6. STRESSES - LARGEST VALUES

CONSTRUCTION ENGINEERING, INC.
 ENGINEERING DEPARTMENT, CHATTANOOGA, TENN.
 DRAWING NO. _____
 DATE 1-27-67 BY RICHARDER
 CHECK DATE 1-27-67 BY CARLOS
 CASE SUPPORT PADS
 NUMBER 5-201-PLA-88
 SHEET 15 OF 31