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5.0 SHIELDING EVALUATION

The NAC-STC uses an optimized multiwall design to provide the most efficient shielding arrangement possible, and to comply with 10 CFR 71 limits. This chapter provides a description of the NAC-STC shield design, design basis contents for the shielding evaluation, and the conservative shielding analyses used to determine the transport dose rates.

The NAC-STC is designed to safely transport spent fuel assemblies in two configurations: directly loaded and canistered. In the directly loaded configuration, standard PWR fuel assemblies are placed directly into a fuel basket installed in the cask cavity. In the canistered configuration, a sealed transportable storage canister loaded with fuel assemblies is placed in an empty cask cavity with top and bottom spacers. In the directly loaded configuration, the NAC-STC can transport up to 26 standard PWR fuel assemblies. In the canistered configuration, the NAC-STC can transport up to 36 Yankee Class fuel assemblies in the Yankee-MPC configuration or up to 26 Connecticut Yankee fuel assemblies in the CY-MPC configuration.

For directly loaded fuel, the shielding evaluation considers reference fuel assemblies in 14×14, 15×15, 16×16 and 17×17 array sizes. The reference fuel assemblies have parameters selected from all of the fuel assemblies of the same array size to maximize the shielding source terms. The design basis fuel for the canistered configuration is the Yankee Class, Combustion Engineering, Type A, 16×16 PWR fuel assembly.

The NAC-STC can also safely transport Greater Than Class C (GTCC) waste in a canistered configuration. The Yankee Class GTCC waste consists primarily of activated steel sections or components, but may also include Zircaloy items. Core baffle sections and dross material are placed in a fuel assembly-sized container, as shown in Figure 5.1-4. Some stainless steel and Zircaloy items may be loaded directly into an interior GTCC loading position. The Connecticut Yankee GTCC waste, also consisting of activated steel, is also placed in a fuel assembly-sized can. The Yankee-MPC and CY-MPC GTCC canisters have 24 loading positions for GTCC waste.

The NAC-STC is assigned a nominal Transport Index of 21 (TI = 21) based on the requirement of 10 CFR 71.4 and the analysis results presented in Section 5.1.4. The maximum dose rate at 1 meter from the NAC-STC in normal conditions of transport is 20.3 mrem per hour, based on the directly loaded reference fuel. The actual measured dose rate is expected to be less.

The shielding evaluation for directly loaded fuel, canistered fuel and GTCC waste demonstrates compliance with 10 CFR 71 limits. The dose rates for both the canistered Yankee Class fuel and GTCC waste, and Connecticut Yankee Class fuel and GTCC waste, are shown to be significantly less than those for the directly loaded fuel configuration for both normal and accident conditions.

The shielding evaluation of the directly loaded configuration is performed using the SAS2H sequence (Hermann, 1995) of the SCALE-4.3 package for the PC (ORNL, 1995). This sequence uses the computer code ORIGEN-S (Hermann, 1989) to calculate the source terms. The MCBEND (AEA Technology, 2000) computer code is used to calculate the cask dose rates for normal transport and hypothetical accident conditions. The shielding analyses show that the dose rates are below regulatory limits.

The shielding evaluation of the Yankee Class canistered fuel and GTCC waste is performed using SCALE 4.3 for the PC (ORNL, 1995). This code uses SAS2H (Herman, 1995) to calculate source terms. One-dimensional shielding evaluations were performed using SAS1 (Knight, 1995). The shielding analyses show that the dose rates are well below the regulatory limits stated in 10 CFR 71 and are well below the dose rates reported for the design basis directly loaded fuel.

The shielding evaluation of the Connecticut Yankee canistered fuel and GTCC waste is performed using the MCBEND Monte Carlo transport code. Fuel source terms are developed using the SCALE isotopics sequence SAS2H (Herman, 1995).

Directly Loaded Fuel

The directly loaded basket construction is based on a tube and disk design. PWR fuel is loaded into 26 fuel tubes fabricated from Type 304 stainless steel sheets. BORAL or TalBor neutron absorber is encased in stainless steel on the outside face of the fuel tube. Twenty 5/8-inch thick aluminum disks are spaced between thirty-three 1/2-inch thick Type 17-4 PH stainless steel support disks to provide heat transfer. Radial shielding of PWR fuel in the directly loaded basket is provided by the multi-wall design of the NAC-STC cask body. Axial shielding is provided by the cask body closure lids and end forgings and the impact limiters.

Yankee Class Canistered Fuel and GTCC Waste

The canister containing Yankee Class fuel or GTCC waste is placed in the NAC-STC cavity with top and bottom spacers. The placement of the canister between the top and bottom spacers

effectively precludes the source regions from streaming through areas above and below the neutron shield and tapered regions of the lead. In addition to the radial and axial shielding provided by the cask body and lids, radial and axial shielding is provided by the canister 5/8-inch shell, the 8 inches of stainless steel from the canister lids and 1 inch of steel from the canister bottom.

The Yankee-MPC fuel basket is of the same design as the steel/aluminum directly loaded basket previously described. It has a shorter overall length to accommodate the dimensions of the design basis Yankee Class fuel, and a smaller diameter to accommodate the inside dimension of the canister. Consistent with these smaller dimensions, the Yankee-MPC basket also has fewer support plates and heat transfer disks than the directly loaded basket.

The Yankee-MPC GTCC basket is a simplified tube-and-disk design. The steel tubes holding the GTCC waste containers are surrounded by a 2.5-inch steel basket support wall and are held in place by steel support disks. Heat transfer disks are not used.

Connecticut Yankee Canistered Fuel and GTCC Waste

The canister containing CY fuel or GTCC waste is placed in the NAC-STC cavity with a bottom spacer. In addition to the radial and axial shielding provided by the cask body and lids, radial and axial shielding is provided by the canister 5/8-inch shell, the 8 inches of stainless steel from the canister lids and 1.75 inches of steel from the canister bottom.

The CY-MPC canistered fuel basket is of the same design as the steel/aluminum directly loaded basket. The basket height is 141.25 inches and has a diameter sized to fit inside the canister, which has an outer diameter of 70.64 inches. The CY-MPC canister has 27 aluminum heat transfer disks and 28 stainless steel support disks.

The CY-MPC GTCC basket is a simplified supported tube design. The steel tubes holding the GTCC waste containers are surrounded by a 1.75-inch thick cylindrical steel basket support wall, which is held in place by steel support ribs.

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5.1 Discussion and Results

The radiation protection provided by the NAC-STC is in the form of solid multi-walled shielding materials, which totally surround the fuel. These shielding materials include steel and lead for gamma shielding and a borated polymer (NS-4-FR) for neutron shielding. The multi-walled arrangement of steel and lead in the NAC-STC provides optimal weight for gamma attenuation. The NS-4-FR neutron shielding material has a hydrogen density close to that of water and serves to moderate fast neutrons which are then captured in the boron. Boron capture in the neutron shield minimizes the contribution of secondary capture gammas to surface dose rates.

The NAC-STC uses a multi-walled arrangement for both radial and axial shields. The arrangement of the radial gamma shielding in the cask body is a 1.5-inch thick stainless steel inner shell and a 2.65-inch thick stainless steel outer shell with a 3.70-inch thick lead filled annulus between them. The radial neutron shield is arranged around the outer steel shell with a 5.5-inch thick NS-4-FR layer which is covered by a 0.25-inch (6 mm) thick neutron shield shell. The bottom of the cask contains a steel/NS-4-FR/steel shield arrangement with the two stainless steel components providing 11.65 inches of gamma shielding and 2 inches of NS-4-FR neutron shielding. The top of the cask has shields in the form of two closure lids. The inner lid also has a steel/NS-4-FR/steel arrangement with 6.0 inches of steel below 2 inches of NS-4-FR and 1.0 inch of steel above it. The outer lid is a 5.25-inch thick steel disk.

5.1.1 Design Criteria

The shielding design criteria for the NAC-STC meets the requirements of 10 CFR 71. For normal conditions, the dose rate limits specified in 10 CFR 71.47 for consignments under exclusive use are: 1,000 mrem/hour on the surface of the enclosed package, 200 mrem/hour on the outer surfaces of transport vehicle and 10 mrem/hour at 2 meters from the vertical planes represented by the outer lateral surfaces of the transport vehicle. The cask surface dose rate is less than 200 mrem/hr, except at the gap between the neutron shield and the upper impact limiter and at the rotation trunnions, where the maximum dose rate is 366 mrem/hr. The maximum dose rate at the personnel barrier, which is the accessible surface of the package, adjacent to the gap between the neutron shield and upper impact limiter, is significantly less than 200 mrem/hr. Note that the cask tie-down structure that is present at this location is conservatively not considered. The 10 mrem/hour criterion has also been met at all locations 2 meters from the railcar. Under hypothetical accident conditions, 10 CFR 71.51 specifies a dose rate limit of

1,000 mrem/hour at 1 meter from the surface of the cask. This criterion has also been met at all locations.

The accessible surface of the package is defined as a personnel barrier that will be on the same plane as the outer radial surface of the top half of the impact limiters. The personnel barrier will attach to the edge of the railcar between the impact limiters. The personnel barrier location is shown in NAC Drawing 423-901.

5.1.2 Design Basis Fuel

The NAC-STC has two configurations for transport of design basis fuel: directly loaded and canistered. The design basis fuel for the directly loaded configuration is described in Section 5.1.2.1. There are two canister configurations. The Yankee-MPC for Yankee Class fuel and GTCC waste and the CY-MPC for Connecticut Yankee Class fuel and GTCC waste. The design basis fuels for shielding for these configurations are described in Sections 5.1.2.2 and 5.1.2.3, respectively.

5.1.2.1 Design Basis Directly Loaded Fuel

The NAC-STC can transport up to 26 directly loaded, intact PWR fuel assemblies over a range of burnups, initial ^{235}U enrichments, and minimum allowable cool times. The general fuel characteristics for directly loaded fuel are given in Table 5.1-1. Detailed material and geometry descriptions for the fuel types evaluated are provided in Section 5.2. Reference fuel assemblies have been developed and analyzed to envelop PWR fuel for 14×14, 15×15, 16×16, and 17×17 array sizes. These assemblies are constructed by surveying assembly data for assemblies less than 165 inches in length (the length of the STC cavity) and using bounding fuel parameters to maximize fuel mass (MTU) and hardware source terms. Decay heats and dose rates have been calculated for a finite range of burnups, initial ^{235}U enrichments, and cool times to generate an allowable loading table, or minimum cool time table. Adherence to the cool timetable ensures that heat load and dose rate limits will not be exceeded.

Three-dimensional dose rates are calculated using a response function methodology. Each of the four fuel assembly array sizes is analyzed over a range of source regions and source types with unit source in each relevant energy group. Source types considered are fuel neutron, fuel gamma

fuel secondary gamma (n-gamma), in-core fuel hardware (grid spacers, steel guide tubes, etc.), plenum, and end fitting hardware. These sources are analyzed in a finite number of energy groups with a unit source in each group. The scalar product of source term and response function allows for the creation of large arrays of dose rate results, whether they are for a single detector, or the maximum or average over a detector surface. In this analysis, detector maximum responses have been used exclusively to generate minimum cool time tables.

5.1.2.2 Design Basis Yankee Class Canistered Fuel and GTCC Waste

The design basis fuel for the Yankee Class canistered configuration for shielding purposes is the Combustion Engineering (CE), Type A, 16 x 16 PWR assembly with an initial enrichment of 3.7 wt % ^{235}U , a uranium mass of 239.4 kilograms, a burnup of 36,000 MWD/MTU and 8.0-year cooling time. To meet maximum cask decay heat limits, an 8.1-year cool time is required. The 8.0-year cooled source terms are conservatively used as the shielding design basis. The dose rates resulting from this assembly are higher than those of the other Yankee Class fuels: CE Type B, Westinghouse, Exxon, and United Nuclear Type A and B fuel assemblies. The design basis Yankee Class fuel characteristics are given in Table 5.1-1. The design basis Yankee Class fuel physical parameters are presented in Table 5.1-2. The design basis canister fuel assembly source terms are presented in Table 5.1-3, and a sketch of the fuel assembly is shown in Figure 5.1-3.

Source terms and dose rate evaluations concluded that the Westinghouse, United Nuclear, and CE Yankee Class fuel assemblies at 32,000 MWD/MTU require minimum cooling times of 22, 11 and 7 years, respectively. The minimum nominal enrichments for these assemblies are 4.94, 4.0 and 3.5 wt % ^{235}U , respectively. Exxon fuel, with a burnup of 36,000 MWD/MTU and a minimum initial enrichment of 3.5 wt % ^{235}U , requires a minimum cooling time of 16 years for assemblies containing steel hardware in the active fuel region, and 10 years for assemblies with Zircaloy hardware. The 10-year cool time for Exxon fuel assemblies with Zircaloy fuel region hardware includes the activation of four hollow Zircaloy replacement rods with stainless steel slugs.

The Yankee-MPC may also hold Reconfigured Fuel Assemblies, Damaged Fuel Cans, or Recaged Fuel Assemblies.

The transportable storage canister may contain one or more Reconfigured Fuel Assemblies. The Reconfigured Fuel Assembly is designed to confine Yankee Class spent fuel rods, or portions thereof, which have been classified as failed. Each assembly can accommodate up to a total of 64 fuel rods. Due to the low number of rods, the reconfigured assembly fuel mass is significantly less than the fuel mass contained in the design basis assemblies described in this section and in Section 5.1.2.1. Because the source term (neutron and gamma) is directly proportional to the fuel mass, for a given burnup, the reconfigured assembly source term is bounded by that of the design basis Yankee Class fuel assemblies. The lower source term of the 64-rod reconfigured assembly more than offsets any reduced self-shielding associated with its lower mass. In addition, each Reconfigured Fuel Assembly fuel rod is placed within a steel enveloping rod. Consequently, a rigorous shielding analysis is not required for the Reconfigured Fuel Assembly.

The Yankee-MPC canister may also contain Damaged Fuel Cans in the four corner basket locations. To accommodate the Damaged Fuel Cans, oversized openings are present in the top and bottom basket weldments, and four 9.3-inch square, 1.4-inch deep recesses are made in the shield lid to accommodate the top fitting. Damaged fuel dose rates are estimated by determining the void area in the intact fuel description and conservatively scaling the source and dose rates to that of a full-density UO_2 region. This approach is conservative since no credit is taken for the increased self-shielding associated with the higher density source region. Only radial damaged fuel dose rates are considered due to the negligible top and bottom axial dose rates calculated for intact fuel. Top and bottom spacers in the cask cavity assure that the Yankee-MPC canister remains centered in the cask cavity.

Minimum cool time for recaged fuel assemblies, assemblies in which United Nuclear fuel rods are moved to a CE skeleton, are conservatively set to the United Nuclear assembly minimum cool time of 13 years. The CE fuel assembly skeleton contains a significantly lower amount of hardware and, therefore, contains a lower potential source term than the United Nuclear fuel assembly skeleton.

The NAC-STC can also safely transport Yankee GTCC waste. The GTCC waste consists primarily of activated steel from the baffle core structure, but includes miscellaneous hardware consisting of sectioned source vanes and fuel assembly cage guide tubes, grid straps and end fittings. Core baffle and dross from cutting operations is placed in a container (see Figure 5.1-4) that is the same size as a Yankee Class fuel assembly. Up to 24 GTCC containers can be loaded into the GTCC canister basket. The miscellaneous hardware material is loaded directly into one

of the interior loading positions of the GTCC basket. The GTCC canister is loaded in the NAC-STC for transport.

The design basis gamma source for the Yankee GTCC waste is determined from dose rate measurements and chemical assay of the GTCC waste. This gamma source is primarily due to the activation of the core baffle from 30 years of neutron flux exposure and to a lesser extent from surface contamination. This source term is used for all of the loading positions of the GTCC basket and bounds the source term for the miscellaneous hardware items. The design basis source term for the GTCC waste canister is 9.493×10^{15} photon/s, which is equivalent to 125,000 curies of ^{60}Co . The design basis thermal output is 1.93 kW.

5.1.2.3 Design Basis Connecticut Yankee Canistered Fuel and GTCC Waste

The design basis Connecticut Yankee (CY) fuels for the shielding evaluation are stainless steel and Zircaloy clad 15 x 15 assemblies. The stainless steel clad assemblies have a maximum burnup of 38,000 MWD/MTU and a minimum of 10-year cool time. The Zircaloy clad assemblies have a maximum burnup of 43,000 MWD/MTU and a minimum of 10-year cool time. The characteristics of the Connecticut Yankee fuel assemblies are presented in Table 5.1-1 and the stainless steel and Zircaloy fuel physical parameters are presented in Table 5.1-2. The fuel assembly source terms are presented in Table 5.1-3. The source regions of the fuel assemblies are shown in Figure 5.1-5.

The NAC-STC can also safely transport Connecticut Yankee GTCC waste. The GTCC waste, consisting of activated steel, is placed in a fuel assembly sized can. Up to 24 GTCC containers can be loaded into the CY-MPC GTCC basket. The CY-MPC GTCC canister is loaded in the NAC-STC cask for transport.

The design basis gamma source for CY GTCC waste is due to the activation of the core baffle during the lifetime of core operation. The source term for the CY-MPC canister is based on 2.77×10^5 curies of ^{60}Co in 14,300 lbs of core baffle material, four years after reactor shutdown.

The CY-MPC may contain up to four reconfigured fuel assemblies positioned in the oversized corner locations in the basket. The CY-MPC reconfigured fuel assembly is designed to confine individual spent fuel rods or portions thereof, within individual stainless steel tubes. Each CY-MPC reconfigured fuel assembly can accommodate up to 100 rods in a 10 x 10 lattice, which

is significantly less than the number of fuel rods in an intact assembly. Because the source term (neutron and gamma) is directly proportional to fuel mass, for a given burnup and enrichment, the source term produced by the fuel rods within the CY-MPC reconfigured assembly is bounded by that of a design basis fuel assembly. Consequently, a rigorous shielding analysis is not required for the CY-MPC reconfigured fuel assemblies.

The CY-MPC may also contain up to four damaged fuel cans positioned in the oversized corner locations of the basket. The CY-MPC damaged fuel can may hold a complete fuel assembly, a lattice or a failed rod storage canister. As such, the shielding analysis conservatively assumes that no damaged fuel cans are present in the canister, as the additional shielding provided by the wall of the can would serve to reduce external dose rates. The effect of damaged fuel migrating into the void space in the upper and lower assembly hardware regions is evaluated explicitly.

The CY-MPC may also contain up to two assemblies with a maximum of two irradiated stainless steel filler rods per assembly. Assemblies with irradiated stainless steel rods may only be loaded in basket positions 13 and 14 as shown in Figure 6.3-3. A dose rate comparison for both normal and accident conditions yields filler rod dose rates are less than 1% of the total dose rate at any surface of the NAC-STC.

5.1.3 Shielding Materials

The shielding materials are selected and arranged to minimize cask weight while maintaining overall shield effectiveness. Lead and steel are chosen as effective gamma radiation shields, and NS-4-FR is provided to efficiently moderate and absorb the neutron radiation, while minimizing the generation of secondary gamma radiation.

5.1.4 Results

For both the directly loaded and the canistered transport configurations, this section demonstrates that the NAC-STC satisfies the regulatory criteria of 10 CFR 71.47 under normal transport condition, and 10 CFR 71.51(a) for hypothetical accident conditions. Specifically, for an exclusive use shipment in an enclosed transport vehicle, the dose rates remain less than 1,000 mrem/hour on the surface of the package, less than 200 mrem/hour at all locations on the surface of the personnel barrier and less than 10 mrem/hour at all locations 2 meters from the edge of the railcar (any point 2 meters from the vertical planes projected from the outer edges of the conveyance). Also, under hypothetical accident conditions, the dose rate is less than 1,000 mrem/hr at 1 meter from the surface of the package. Therefore, the NAC-STC satisfies the shielding criteria of 10 CFR 71.

5.1.4.1 Results of the Shielding Evaluation for Directly Loaded Fuel

The maximum dose rates calculated for the normal transport conditions are shown in Table 5.1-4, with locations of the maximum dose rates shown in Figure 5.1-2. Cask surface dose rates do not exceed the regulatory limit for a closed transport vehicle of 1,000 mrem/hour at the surface of the package. The dose rates at 2 meters from the railcar comply with the 10 mrem/hour regulatory limit.

The maximum normal conditions surface dose rate at the cask radial midplane is 41 mrem/hour. The highest dose rate, occurring on the surface of the cask at the gap between the radial neutron shield and the upper impact limiter, is 366.4 mrem/hour. All cask surface dose rates are much less than 1,000 mrem/hour. Ducting of neutrons through the copper/stainless steel fins is considered in Section 5.4.1.1. The results of the ducting evaluation show that this phenomenon has a very small effect on the total cask dose rate. Azimuthal variations in the calculated dose rate are considered in the explicit heat fin and neutron shield model. The neutron dose rate increase resulting from the ducting is offset by the reduction of the gamma dose rate resulting from the additional shielding provided by the fins.

Table 5.1-5 provides accident dose rates that could occur in the event of the loss of all gaseous elements in the neutron shield combined with radial and axial lead slumps due to cask side and end drops. Although the neutron shield material exceeds its safe operating temperature limits in the fire accident, a complete loss of neutron shielding is not credible for the NAC-STC. Some of the neutron shielding capability may be lost, however, as a result of the fire accident. Therefore, the accident shielding calculations conservatively assume a complete loss of gaseous elements in the neutron shielding. In the event of a cask end drop, it is possible for the lead gamma shielding to slump and fill the annular gap (if one exists) created by the cooling of the lead after fabrication. For worst case conditions, this accident could create a 2.35-inch gap at the top or bottom of the lead annulus. If the cask is subject to a side drop, the lead gamma shielding could slump and create a void on the upper side of the cask. An evaluation of this accident shows the lead thickness may be reduced by a maximum of 0.928-inch. The dose rates shown in Table 5.1-5 show that neither the loss of the neutron shielding nor the slumping of the lead will result in a dose rate that exceeds the hypothetical accident dose rate limit of 1,000 mrem/hour at 1 meter from the cask surface.

Therefore, the NAC-STC fulfills the design criteria of Chapter 1 in that under normal transport conditions, the maximum dose rates are less than 1,000 mrem/hour on the surface of the package, less than 200 mrem/hour at all locations at the surface of the personnel barrier, and less than 10 mrem/hour at all locations 2 meters from the personnel barrier. The cask also satisfies the hypothetical accident criteria of 1,000 mrem/hour at 1 meter from the cask surface.

5.1.4.2 Shielding Evaluation for Yankee Class Canistered Fuel and GTCC Waste

A 1-D radial and axial shielding analysis was performed for both the canistered Yankee Class fuel and GTCC waste under normal and hypothetical accident conditions. The dose rates for canistered fuel (Combustion Engineering, 36,000 MWD/MTU, 8-year cooled) are provided in Tables 5.1-6 and 5.1-7. These dose rates are provided in Tables 5.1-8 and 5.1-9 for the GTCC waste. Under normal conditions, the canister is positioned in the cavity with top and bottom spacers, and the impact limiters are in place on the cask. Under accident conditions (i.e., 30-foot drop and fire accident), the radial midplane results assume loss of neutron shielding. A complete loss of neutron shielding is not credible for the NAC-STC. However, because of the elevated fire accident temperatures, the neutron shields exceed their safe operating limits and some neutron shielding capability may be lost. Also, in the axial models, it is assumed that the cavity spacers are crushed, the impact limiters are lost, and the canister is positioned at either the top or the bottom of the cavity.

The maximum calculated dose at the surface of the cask centerline when loaded with canistered Yankee Class fuel under normal conditions is 10.25 mrem/hour. This is much less than the 41 mrem/hour for the same location with the directly loaded reference fuel in the cask. In the accident condition involving loss of neutron shielding and lead slump, a maximum dose rate of 262.76 mrem/hour is calculated at 1 meter from the radial midplane of the NAC-STC. Based on the 4.6 dose multiplier assigned to the damaged fuel region in Section 5.3.2.3, the maximum damaged fuel dose rate is 7.91 mrem/hr at 2 meters and 47 mrem/hr on the cask surface. The multiplier conservatively does not account for self-shielding of the increased source mass or the presence of only four damaged fuel cans in the 36-assembly basket. For the accident case, the 1 meter dose rate for damaged fuel is 18.9 mrem/hr. Consequently, the Transport Index of 21 remains controlled by the directly loaded contents condition. This is also less than the directly loaded reference fuel accident dose rates shown in Table 5.1-5 and is well below 10 CFR 71 regulatory limits.

The maximum calculated dose at the surface of the cask centerline when loaded with GTCC waste under normal conditions of transport is 7.03 mrem/hour. This is much less than the 41 mrem/hour for the same location with the directly loaded reference fuel in the cask. In the accident condition, a maximum dose rate of 55.77 mrem/hour is calculated at 1 meter from the radial surface of the NAC-STC. This is also much less than the directly loaded reference fuel accident dose rates shown in Table 5.1-5 and is well below 10 CFR 71 regulatory limits.

5.1.4.3 Shielding Evaluation for Connecticut Yankee Class Fuel and GTCC Waste

A three-dimensional radial and axial shielding analysis was performed for both the Connecticut Yankee fuel and GTCC waste under normal and hypothetical accident conditions. The dose rates for canistered fuel (stainless steel clad, 38,000 MWD/MTU, 10-year cooled and Zircaloy clad, 43,000 MWD/MTU, 10-year cooled) are provided in Tables 5.1-10 through 5.1-13. The dose rates for the GTCC waste are provided in Tables 5.1-14 and 5.1-15.

Under normal conditions, the canister is positioned in the canister cavity with a bottom spacer and the impact limiters are in place on the cask. Under accident conditions (i.e., 30-foot drop and fire accident), the NAC-STC is modeled without a radial neutron shield and without the impact limiters. A combined slump of the lead shielding is assumed to be present radially and at both the top and bottom axial locations. Spacer deformation does not occur and is not modeled.

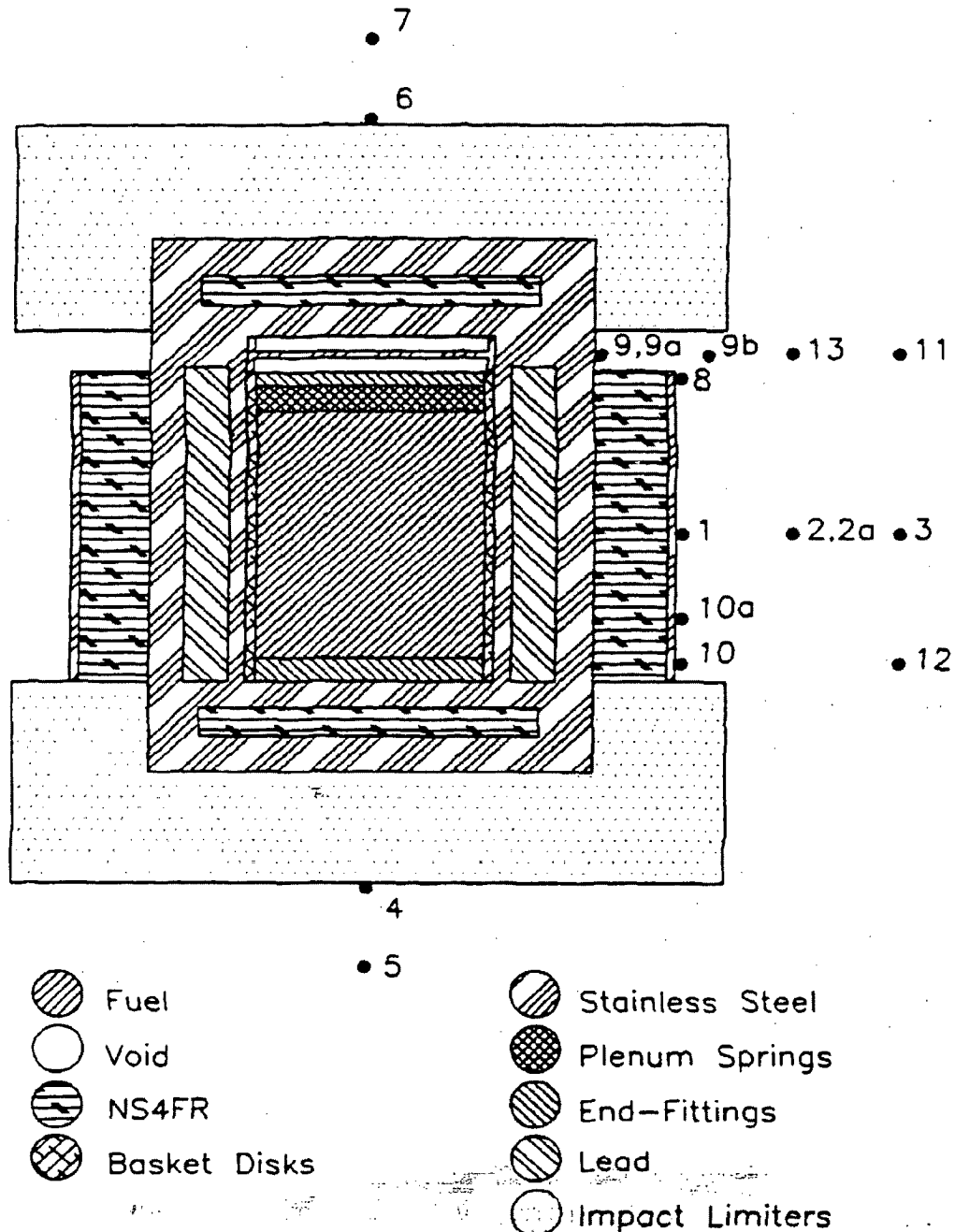
The maximum calculated dose at the radial surface of the cask with CY-MPC fuel under normal conditions is 49.1 mrem/hr. In the accident condition involving loss of neutron shielding, a maximum dose rate of 369 mrem/hr is calculated at 1 meter from the radial midplane of the NAC-STC. This is much less than the directly loaded reference fuel accident dose rates shown in Table 5.1-5 and is well below 10 CFR 71 regulatory limits.

The maximum calculated dose at the radial surface of the cask when loaded with CY-MPC GTCC waste under normal conditions of transport is 6.1 mrem/hr. This is much less than the 41 mrem/hr at the cask centerline with the directly loaded reference fuel in the cask. In the accident condition, a maximum dose rate of 25.0 mrem/hr is calculated at 1 meter from the radial surface of the NAC-STC. This is also much less than the directly loaded reference fuel fire accident dose rates shown in Table 5.1-5 and is well below 10 CFR 71 regulatory limits.

Locations of the maximum dose rates for CY-MPC fuel and GTCC waste are shown in Figures 5.1-6 and 5.1-7 for normal conditions of transport and the accident condition, respectively.

The dose rate increase due to the inclusion of up to two assemblies with a maximum of two irradiated stainless steel filler rods per assembly is less than 1% of the total dose rate at any surface of the NAC-STC.

Figure 5.1-1 Detector Locations for Yankee Class Canistered Fuel and GTCC Waste



Detector locations are described in Tables 5.1-6 through 5.1-9

Figure 5.1-2 Maximum Dose Rate Locations for the Three-Dimensional Directly Loaded Fuel
Analysis in Normal Conditions

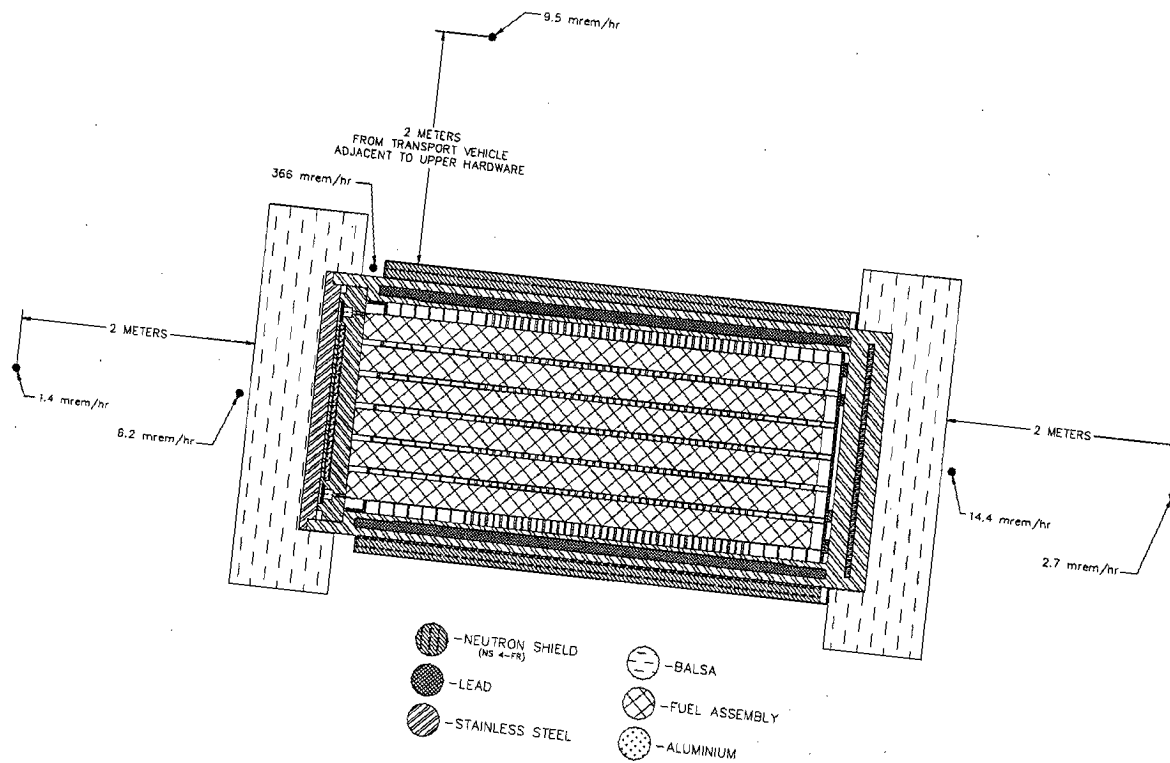


Figure 5.1-3 Design Basis Yankee Class Combustion Engineering Fuel Assembly

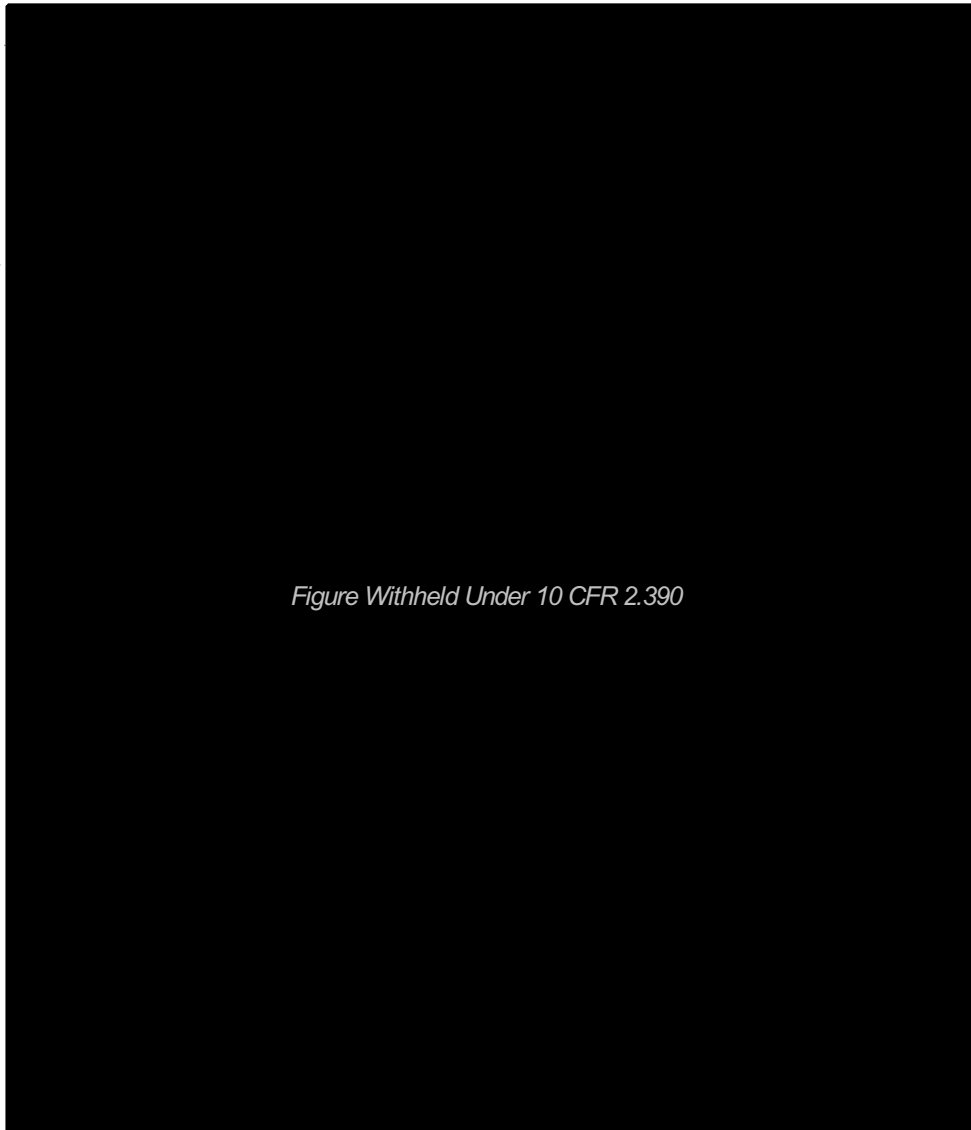


Figure 5.1-4 Yankee GTCC Waste Container

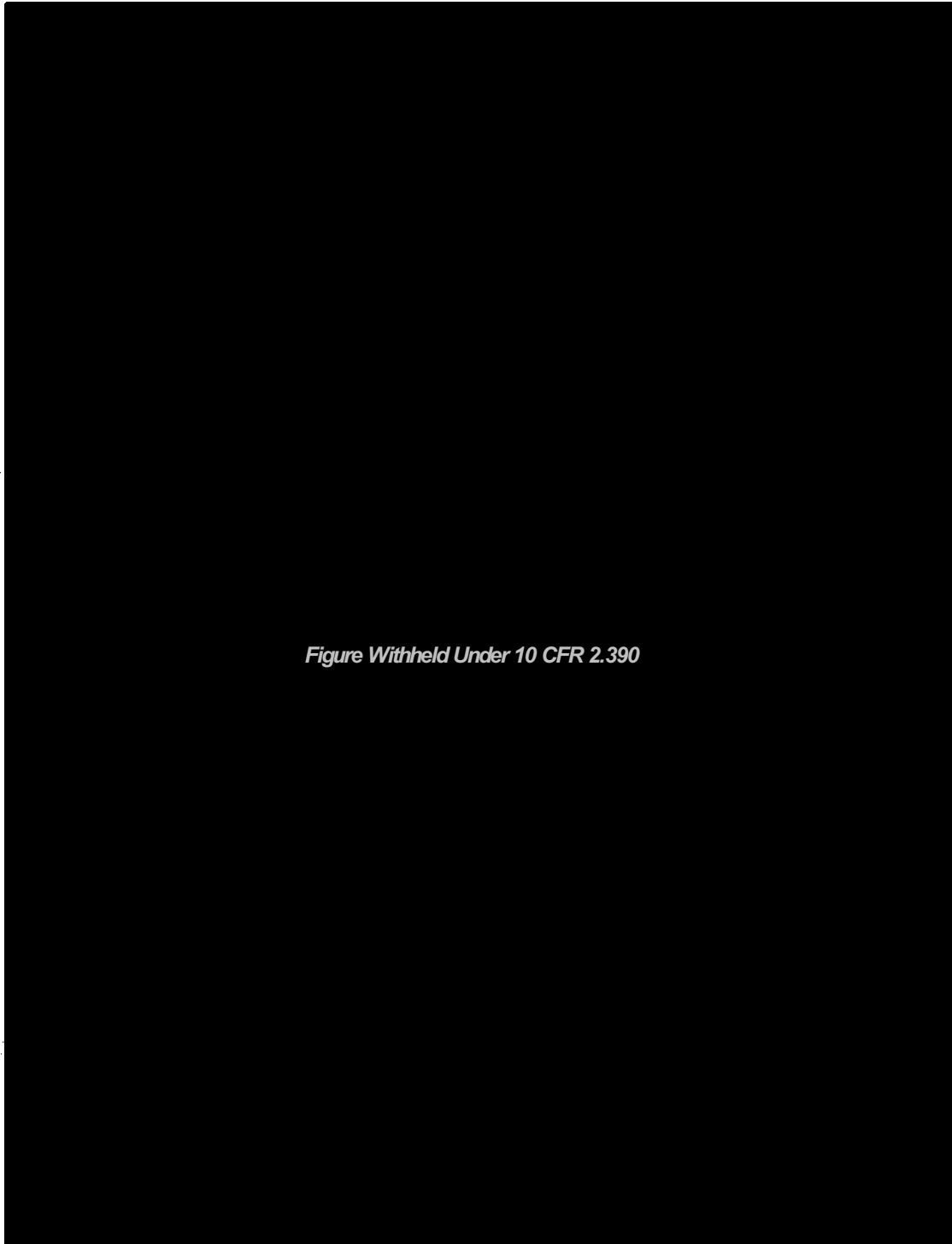


Figure Withheld Under 10 CFR 2.390

Figure 5.1-5 Connecticut Yankee Design Basis Fuel Assembly Source Regions and Elevations



(Dimensions in cm)

Figure 5.1-6 Location of Maximum Dose Rates for CY-MPC Fuel and GTCC Waste in Normal Conditions of Transport

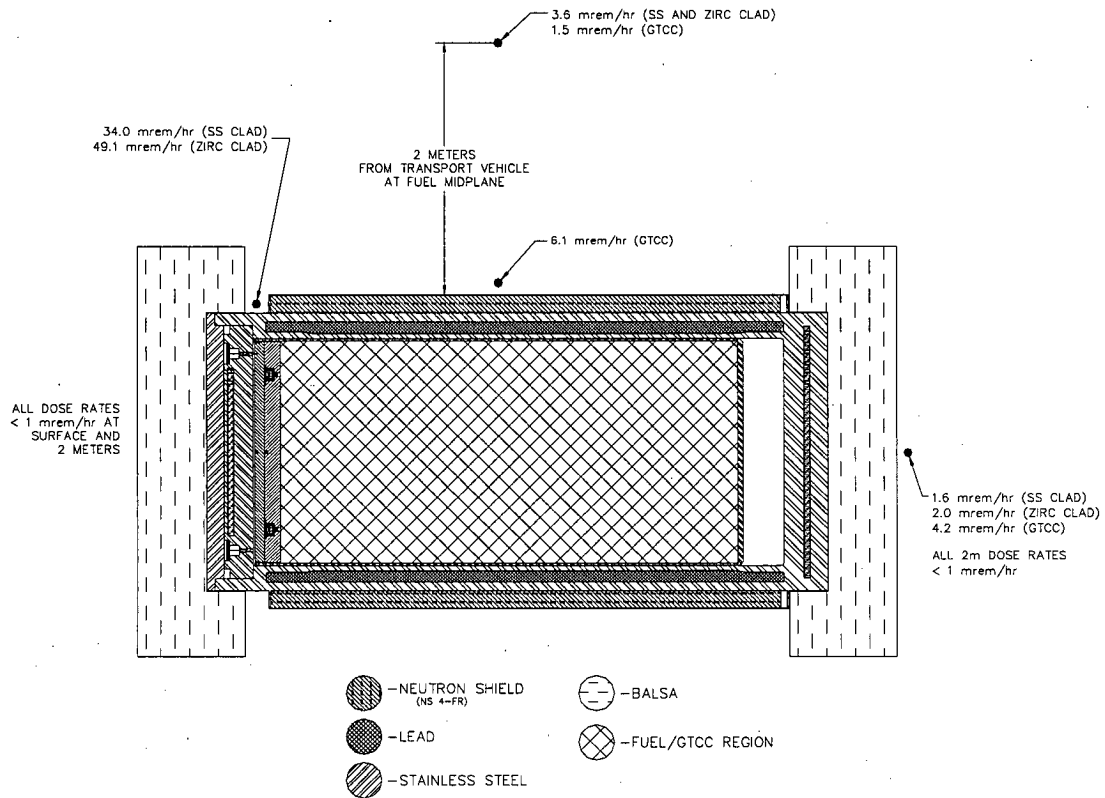


Figure 5.1-7 Location of Maximum Dose Rates for CY-MPC Fuel and GTCC Waste in Accident Conditions

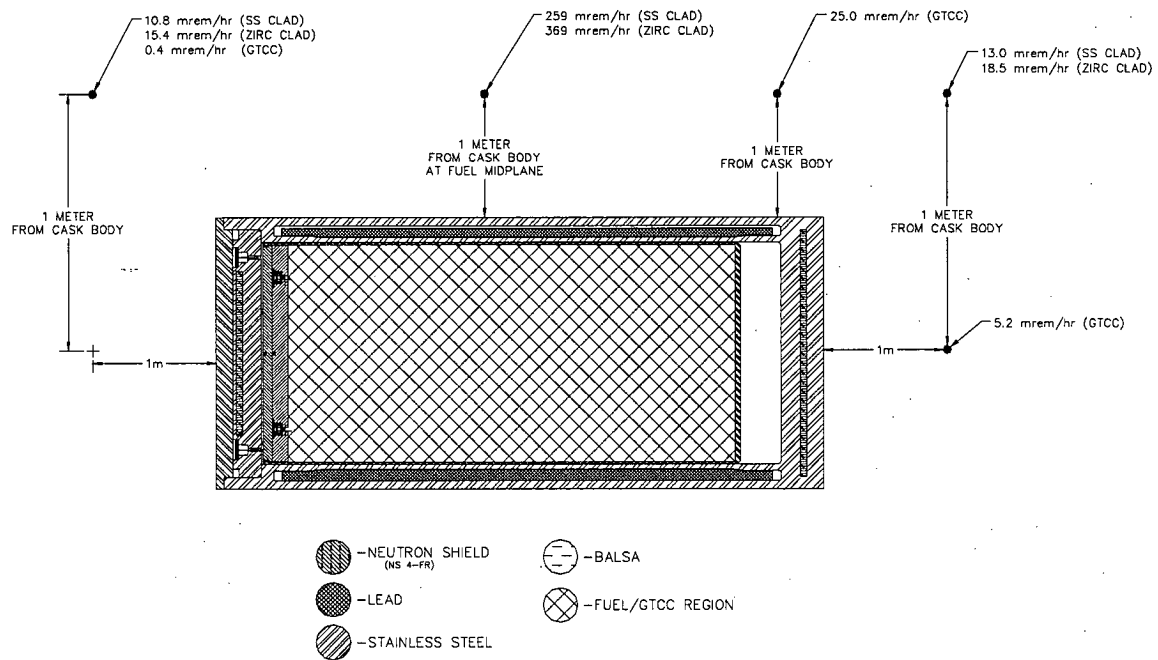


Table 5.1-1 Type, Form, Quantity and Potential Sources of the Fuel Used for Design Basis Directly Loaded and Canistered Fuel

	Design Basis Directly Loaded Fuel	Design Basis Yankee Class Canistered Fuel
Fuel Type and Parameters	<ul style="list-style-type: none"> • PWR, 14 x 14, 15 x 15, 16 x 16 and 17 x 17 • Array-dependent maximum uranium mass • Variable minimum initial ^{235}U enrichment • 45,000 MWD/MTU maximum burnup • 0.85 kW per assembly maximum decay heat, 22.1 kW per cask for 26 assemblies • Variable minimum cool time 	<ul style="list-style-type: none"> • Yankee Class PWR Combustion Engineering, 16 x 16 Type A • 239.4 kg maximum uranium mass • 3.7 wt % maximum initial ^{235}U enrichment¹ • 36,000 MWD/MTU maximum burnup² • 0.347 kW per assembly maximum decay heat, 12.5 kW per cask for 36 assemblies • 8.1 years (or more) decay time after reactor discharge²
Fuel form	Intact assemblies	Yankee Class Fuel
Quantity	26 design basis fuel assemblies	36 design basis fuel assemblies
Heat Load	22.1 kilowatts, thermal per cask	12.5 kilowatts, thermal per cask
Sources of Fuel	Commercial PWR nuclear power reactors	Commercial Yankee Class nuclear power reactors

1. 3.7 wt % ^{235}U is used for the 36,000 MWD/MTU fuel assembly shielding source terms. It yields higher source terms than the 3.9 wt % ^{235}U used in the criticality analysis.
2. Yankee Class Westinghouse, United Nuclear and Combustion Engineering (3.5 wt % ^{235}U) fuel assemblies with burnups up to 32,000 MWD/MTU require minimum cool times of 22, 11 and 7 years, respectively. Exxon assemblies with burnups up to 36,000 MWD/MTU require a minimum cool time of 16 years for assemblies containing steel hardware in the active fuel region and 10 years for assemblies with Zircaloy hardware.

Table 5.1-1 Type, Form, Quantity and Potential Sources of the Fuel Used for Design Basis Directly Loaded and Canistered Fuel (Continued)

	Design Basis Stainless Steel Clad Connecticut Yankee Canistered Fuel	Design Basis Zircaloy Clad Connecticut Yankee Canistered Fuel
Fuel Type and Parameters	<ul style="list-style-type: none"> • Connecticut Yankee 15 x 15 Stainless Steel Clad • 431.7 kg maximum uranium mass • 3.65 wt % maximum initial ^{235}U enrichment • 38,000 MWD/MTU maximum burnup • 10 years decay time after reactor discharge 	<ul style="list-style-type: none"> • Connecticut Yankee 15 x 15 Zircaloy Clad • 395.2 kg maximum uranium mass • 3.59 wt % maximum initial ^{235}U enrichment • 43,000 MWD/MTU maximum burnup • 10 years decay time after reactor discharge
Fuel form	Intact or damaged assemblies	Intact or damaged assemblies
Quantity	26 design basis fuel assemblies	26 design basis fuel assemblies
Heat Load	15.6 kilowatts, thermal per cask	16.3 kilowatts, thermal per cask
Sources of Fuel	Connecticut Yankee Haddam Neck nuclear reactor	Connecticut Yankee Haddam Neck nuclear reactor

Table 5.1-2 Design Basis Canistered Fuel - Physical Parameters

PARAMETER	Canistered Fuel		
	Yankee Class	CY Stainless Steel Clad	CY Zircaloy Clad
Assembly Rod Array	16 x 16	15 x 15	15 x 15
Assembly Length, in	111.79	137.1	137.1
Active Fuel Length, in	91	121.8	121.1
No. of Fuel Rods	231	204	204
Rod Pitch, in	0.472	0.563	0.563
Cladding Material	Zircaloy-4	Stainless Steel	Zircaloy-4
Rod Diameter, in	0.365	0.422	0.424
Cladding Thickness, in	0.024	0.0165	0.025
Pellet Diameter, in	0.3105	0.3835	0.3680
Pellet Material	UO ₂ (sintered)	UO ₂ (sintered)	UO ₂ (sintered)
Maximum Fuel Rod Pressure, psig	315	475	475
Theoretical Density, percent	95	95	95
Maximum Initial Enrichment, wt % U ²³⁵	3.9 ²	4.03 ⁴	4.61 ⁴
Design Basis Burnup, MWD/MTU	36,000	38,000	43,000
Weight of U, kg (typical)	239.4 ¹	421.2	386.7
Weight of UO ₂ , kg (typical)	271.6	477.8	438.7
Upper End-Fitting, kg/assembly	5.5	11.24	11.84
Lower End-Fitting, kg/assembly	5.18	8.85	5.44
Upper Plenum Springs, kg/assembly	0.762	- ³	- ³
Upper Plenum Grid, kg/assembly	0.590	3.879 ³	5.137 ³
Lower Plenum Grid, kg/assembly	NA	NA	NA

- 1 A bounding value of 469 kg is used for the total mass of Uranium.
- 2 An initial enrichment of 3.7 wt % ²³⁵U is used in source term generation to maximize the neutron source.
- 3 Upper plenum spring and grid hardware mass is combined.
- 4 Initial enrichments of 3.65 wt % ²³⁵U and 3.59 wt % ²³⁵U for stainless steel and Zircaloy clad fuels, respectively, to maximize the neutron sources.

Table 5.1-3 Nuclear Parameters of the Canistered Fuels and GTCC Waste

Configuration	Yankee Class Fuel	Yankee GTCC Waste ¹	CY Stainless Steel Clad Fuel	CY Zircaloy Clad Fuel	CY GTCC Waste
No. of Fuel Assemblies or Containers	36	24	26	26	24
Burnup, MWD/MTU	36,000	N/A	38,000	43,000	N/A
Cooling Time, years	8	N/A	10	10	10
Gamma Source, MeV/s photons/sec	2.856×10^{16} 6.423×10^{16}	1.16×10^{16} 9.493×10^{15}	7.712×10^{16}	8.002×10^{16}	1.81×10^{16}
Neutron Source, neutrons/sec	2.415×10^9	N/A	3.689×10^9	5.348×10^9	N/A
Core Grids, photons/sec ²	0.0	N/A	3.879×10^{15}	1.974×10^{15}	N/A
Upper end-fitting ⁶⁰ Co Source, photons/s ³	8.330×10^{13}	N/A	1.018×10^{14}	1.204×10^{14}	N/A
Lower end-fitting ⁶⁰ Co Source, photons/sec ³	7.876×10^{13}	N/A	9.926×10^{13}	5.532×10^{13}	N/A
Upper Plenum Hardware ⁶⁰ Co Source, photons/sec ³	2.309×10^{13}	N/A	8.937×10^{13}	1.045×10^{14}	N/A
Lower Plenum ⁶⁰ Co Source, photons/sec	5.242×10^{13}	N/A	N/A	N/A	N/A

1 Includes depleted Sb-Be source vanes, fuel assembly cage components and core baffle steel.

2 CY hardware sources include steel guide tubes for Zircaloy clad fuel and steel guide tubes and steel clad for stainless steel clad fuel.

3 CY upper end-fitting, lower end-fitting, and upper plenum source strengths are total sources, not ⁶⁰Co sources.

Table 5.1-4 Directly Loaded Fuel Maximum Dose Rates for Normal Conditions of Transport

Detector	Source	Surface		2 meter ¹	
		mrem/hr	RSD	mrem/hr ²	RSD
Top Axial	Neutron	0.5	0.2%	0.2	0.2%
	Gamma	5.7	0.4%	1.2	0.4%
	Total	6.2	0.3%	1.4	0.3%
Radial	Neutron	152.2	0.3%	2.9	0.3%
	Gamma	214.2	0.3%	6.6	0.4%
	Total	366.4	0.2%	9.5	0.3%
Bottom Axial	Neutron	4.0	0.3%	0.8	0.2%
	Gamma	10.4	0.6%	1.9	0.9%
	Total	14.4	0.4%	2.7	0.7%

1. Dose rates are rounded to the indicated precision.
2. Dose rates at 2 meter locations radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.1-5 Directly Loaded Fuel Maximum Dose Rates for Hypothetical Accident Conditions

Detector	Source	Surface ¹		1 meter ¹	
		mrem/hr ²	RSD	mrem/hr ²	RSD
Top Axial	Neutron	31.3	0.5%	23.4	0.9%
	Gamma	20.5	0.8%	10.9	2.7%
	Total	51.8	0.5%	34.3	1.1%
Radial ³	Neutron	1586	0.2%	578	0.2%
	Gamma	52	6.6%	27	5.1%
	Total	1638	0.3%	605	0.3%
Bottom Axial	Neutron	119.3	0.3%	51.2	6.3%
	Gamma	67.2	0.8%	17.9	0.9%
	Total	186.5	0.3%	69.1	4.7%

1. The hypothetical accident conditions include a loss of all oxygen, hydrogen, and nitrogen in the radial neutron shield material and radial and axial lead slumps.
2. Dose rates are rounded to the indicated precision.
3. The azimuthal maximum radial dose rates are 1729 (1.9%) and 665 (4.4%) mrem/hr at the surface and at 1 meter from the surface, respectively.

Table 5.1-6 Combined Top, Radial Midplane, and Bottom Canistered Yankee Class Fuel
Dose Rates for Normal Conditions of Transport

Location	Detector I.D.	Radiation	Dose Rate (mrem/hr)
Radial Surface, fuel midplane	1	Fuel Gamma	3.89
		Fuel Neutron	3.46
		(n,γ)	<u>2.90</u>
		TOTAL	10.25
Radial, 1m from cask surface, fuel midplane	2	Fuel Gamma	1.73
		Fuel Neutron	1.29
		(n,γ)	<u>1.09</u>
		TOTAL	4.11
Radial, 2m from transport vehicle, fuel midplane**	3	Fuel Gamma	0.79
		Fuel Neutron	0.52
		(n,γ)	<u>0.41</u>
		TOTAL	1.72
Bottom impact limiter surface, axial centerline	4	Fuel Gamma	0.09
		Upper Plenum Gamma	0.13
		Top Endfitting Gamma	0.37
		Fuel Neutron	0.01
		(n,γ)	<u>0.04</u>
Bottom, 2m from surface of impact limiter, axial centerline	5	TOTAL	0.64
		Fuel Gamma	0.05
		Upper Plenum Gamma	0.07
		Top Endfitting Gamma	0.00*
		Fuel Neutron	0.00*
		(n,γ)	<u>0.02</u>
Top impact limiter surface, axial centerline	6	TOTAL	0.14
		Fuel Gamma	0.00*
		Upper Plenum Gamma	0.00*
		Top Endfitting Gamma	0.00*
		Fuel Neutron	0.00*
		(n,γ)	<u>0.00*</u>
Top, 2m from surface of impact limiter, axial centerline	7	TOTAL	0.00
		Fuel Gamma	0.00*
		Upper Plenum Gamma	0.00*
		Top Endfitting Gamma	0.00*
		Fuel Neutron	0.00*
		(n,γ)	<u>0.00*</u>
		TOTAL	0.00

* Values are less than 0.005.

** A neutron peaking factor of $(1.2)^{4.2} = 2.15$ is applied to the detector locations at or near the radial midplane of the fuel region.

Table 5.1-7 Combined Top, Radial Midplane, and Bottom Canistered Yankee Class Fuel
Dose Rates for Hypothetical Accident Conditions

Location	Detector I.D.	Radiation	Dose Rate (mrem/hr)
Radial, 1m from cask surface, fuel midplane without neutron shield ¹	2a	Fuel Gamma	32.11 ⁴
		Fuel Neutron	230.14
		(n, γ)	<u>0.50</u>
		TOTAL	262.76
Bottom, 1m from cask surface, axial centerline, without neutron shield (assumes loss of impact limiter) ^{1,2}	4	Fuel Gamma	0.81
		Upper Plenum Gamma	1.35
		Top Endfitting Gamma	4.04
		Fuel Neutron	5.35
		(n, γ)	<u>0.10</u>
		TOTAL	11.65
Top, 1m from cask surface, axial centerline, without neutron shield (assumes loss of impact limiter) ²	6	Fuel Gamma	0.00 ³
		Upper Plenum Gamma	0.00 ³
		Top Endfitting Gamma	0.00 ³
		Fuel Neutron	18.20
		(n, γ)	<u>0.01</u>
		TOTAL	18.25

1 Assumes complete loss of neutron shielding material.

2 Assumes loss of impact limiters and positioning of the canister either top or bottom of cavity.

3 Values are less than 0.005.

4 Assumes 0.88 inch reduction in lead shielding due to side drop lead slump.

Table 5.1-8 Canistered Yankee GTCC Waste Dose Rates for Normal Conditions of Transport

Location	Detector I.D.	Radiation	Dose Rate(mrem/hr)
Radial Surface, fuel midplane	1	Neutron	0.00
		Gamma	<u>7.03</u>
		Total	7.03
Radial, 1m from cask surface, fuel midplane	2	Neutron	0.00
		Gamma	<u>3.17</u>
		Total	3.17
Radial, 2m from transport vehicle, midplane	3	Neutron	0.00
		Gamma	<u>1.49</u>
		Total	1.49
Top impact limiter surface, axial centerline	6	Neutron	0.00
		Gamma	<u>0.00</u>
		Total	0.0
Top 2m from impact limiter surface, axial centerline	7	Neutron	0.00
		Gamma	<u>0.00</u>
		Total	0.0
Bottom impact limiter surface, axial centerline	4	Neutron	0.0
		Gamma	2.54
		Total	2.54
Bottom, 2m from cask surface, axial centerline	5	Neutron	0.00
		Gamma	<u>0.46</u>
		Total	0.46

Table 5.1-9 Canistered Yankee GTCC Waste Dose Rates for Hypothetical Accident Conditions

Location	Detector I.D.	Dose Rate (mrem/hr)
Radial, 1m from cask surface, fuel midplane, without neutron shielding ¹	2a	55.77
Top surface 1m from cask surface, axial centerline ²	6	0.01
Bottom, 1m from cask surface, axial centerline, without neutron shield ^{2,3}	4	22.88

- 1 Assumes complete loss of neutron shielding material and lead slump. Loss of neutron shielding alone results in a dose of 12.15 mrem/hr. This dose is increased by a factor of 4.59 to account for a 0.88 inch reduction in lead thickness due to lead slump.
- 2 Assumes loss of impact limiters and positioning of the canister in either the top or bottom of the cavity.
- 3 Assumes complete loss of neutron shielding material.

Table 5.1-10 Connecticut Yankee Stainless Steel Clad Fuel Maximum Dose Rates for Normal Conditions of Transport

Detector	Source	Surface		2 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Neutron	0.1	0.5%	0.1	2.8%
	Gamma	0.1	4.9%	0.1	3.0%
	Total	0.3	2.4%	0.1	2.1%
Radial	Neutron	30.2	0.5%	0.9	0.6%
	Gamma	3.8	2.2%	2.7	1.4%
	Total	34.0	0.5%	3.6	1.0%
Bottom Axial	Neutron	0.5	0.4%	0.1	1.2%
	Gamma	1.2	0.6%	0.2	0.5%
	Total	1.6	0.4%	0.3	0.5%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.1-11 Connecticut Yankee Zircaloy Clad Fuel Maximum Dose Rates for Normal Conditions of Transport

Detector	Source	Surface		2 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Neutron	0.2	0.7%	0.1	3.7%
	Gamma	0.2	2.0%	0.1	2.1%
	Total	0.4	1.0%	0.2	2.2%
Radial	Neutron	43.6	0.4%	1.3	0.5%
	Gamma	5.5	5.5%	2.3	1.2%
	Total	49.1	0.7%	3.6	0.8%
Bottom Axial	Neutron	0.8	0.4%	0.1	0.9%
	Gamma	1.2	2.9%	0.2	1.9%
	Total	2.0	1.8%	0.3	1.1%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.1-12 Connecticut Yankee Stainless Steel Clad Fuel Maximum Dose Rates for Hypothetical Accident Conditions

Detector	Source	Surface		1 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Neutron	2.1	0.6%	9.5	1.7%
	Gamma	0.2	11.8%	1.2	2.5%
	Total	2.4	1.3%	10.8	1.5%
Radial	Neutron	746	0.6%	234	0.4%
	Gamma	60	8.7%	25	1.3%
	Total	806	0.9%	259	0.4%
Bottom Axial	Neutron	4.7	0.5%	12.3	3.2%
	Gamma	3.8	0.7%	0.8	9.9%
	Total	8.5	0.4%	13.0	3.1%

Table 5.1-13 Connecticut Yankee Zircaloy Clad Fuel Maximum Dose Rates for Hypothetical Accident Conditions

Detector	Source	Surface		1 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Neutron	3.1	0.6%	14.1	1.7%
	Gamma	0.3	5.4%	1.3	1.3%
	Total	3.4	0.8%	15.4	1.6%
Radial	Neutron	1,123	0.6%	348	0.4%
	Gamma	47	3.9%	21	1.8%
	Total	1,170	0.6%	369	0.4%
Bottom Axial	Neutron	7.6	0.5%	17.6	3.4%
	Gamma	3.7	1.8%	0.9	14.2%
	Total	11.3	1.3%	18.5	3.3%

Table 5.1-14 Connecticut Yankee GTCC Waste Maximum Dose Rates for Normal Conditions of Transport

Detector	Surface		2 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	< 0.1	1.2%	< 0.1	1.4%
Radial	6.1	0.6%	1.5	0.3%
Bottom Axial	4.2	0.2%	0.7	0.3%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.1-15 Connecticut Yankee GTCC Waste Maximum Dose Rates for Hypothetical Accident Conditions

Detector	Surface		1 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	< 0.1	8.5%	0.4	11.4%
Radial	141.4	10.8%	25.0	5.7%
Bottom Axial	16.7	0.2%	5.2	0.2%

5.2 Source Specification

This section presents the source specifications for the directly loaded fuel and for the Yankee-MPC and CY-MPC fuel and GTCC waste configurations.

5.2.1 Directly Loaded Fuel Source Specification

The directly loaded NAC-STC is designed to safely transport a range of 14×14, 15×15, 16×16, and 17×17 fuel assemblies. The analyzed fuel assemblies are reference fuel assemblies, with assembly geometry and activated hardware masses chosen to maximum uranium loading (MTU) and activated hardware source term.

In order to generate a minimum cool time table for directly loaded fuel, each fuel assembly is analyzed over a range of burnups, initial ^{235}U enrichments and cool times. Fuel assembly burnup is evaluated from 30,000 MWD/MTU to 45,000 MWD/MTU in 5,000 MWD/MTU increments. Initial ^{235}U enrichments are evaluated from 1.7 to 4.5 wt % ^{235}U in 0.2 wt % increments. Cool times range from 5 to 40 years with varying increments. This matrix creates a total of 1,080 source terms for each assembly (4 burnups × 15 enrichments × 18 cool times).

Neutron and gamma source terms for the directly loaded design basis fuel are calculated with the ORIGEN-S computer code (Hermann, 1989) as part of the SAS2H sequence (Hermann, 1995) in the SCALE 4.3 code package for the PC (ORNL, 1995). ORIGEN-S also calculates the gamma spectrum, the neutron spectrum, and the concentration of radiologically important isotopes such as ^3H , ^{131}Xe , ^{129}I , ^{85}Kr , ^{134}Cs , ^{137}Cs and ^{60}Co . Reactor operating conditions assumed for the analysis are shown in Table 5.2-2. The SAS2H-generated source spectra are rebinned onto the standard 28 group neutron and 22 group gamma scheme used in MCBEND as shown in Tables 5.2-21 and 5.2-22, respectively. Source terms are generated for the fuel and fuel assembly hardware. The hardware activation is calculated by light element transmutation using the in-core neutron flux spectrum produced by the SAS2H neutronics model.

The fuel-dependent input data for the shielding and source term evaluations of directly loaded design basis PWR assemblies are given in Table 5.2-1. Fuel assembly parameters have been selected to maximize fuel mass and, therefore, fuel source terms. Fuel assembly hardware masses have likewise been selected to maximize hardware source term.

Fuel assembly parameters used in the SAS2H source term analysis and the MCBEND shielding analysis are identical. Parameters necessary to generate SAS2H input are shown in Table 5.2-3.

Fuel neutron, fuel gamma, and hardware gamma radiation contribute at varying levels to cask dose rates due to significant changes in the material composition of the cask shields at different radial and axial locations. As such, no single source term produces a bounding set of dose rates at all locations. For example, the radial maximum surface dose rate (normal conditions) of 366 mrem/hr is produced by a 40,000 MWD/MTU, 2.3 wt % ^{235}U , 10-year cooled source term, shown in Table 5.2-6. Top axial maximum dose rates (normal conditions) are produced by a 30,000 MWD/MTU, 2.3 wt % ^{235}U , 6-year cooled source term. By employing the response function method to calculate maximum dose rates, the limiting source term becomes a result of the analysis, rather than an input, and the limiting source term and dose rate are captured for radial and axial detectors and normal and accident conditions.

The end-fitting, plenum spring and grid spacer activations are calculated by ORIGEN-S using the same burnup cycle as the fuel. The fuel hardware masses activated are provided in Table 5.2-4 for the directly loaded fuel. The grid spacers and other fuel hardware in the core region are conservatively assumed to be exposed to 100 percent of the flux in the core. For the plenum springs, the grid spacers in the plenum region, and the bottom end-fittings, 20 percent of the flux in the core is used for irradiation purposes. For the top end-fittings, 10 percent of the flux in the core is used for irradiation purposes. These irradiation values are taken from Luksic. The amount of ^{59}Co present in the grid spacers and end-fittings was taken as 1.2 gram per kilogram of material, irrespective of being Inconel or Type 304 stainless steel. However, the value is conservative for both stainless steel and Inconel, as most nuclear-grade material specifications require less than 1 gram of ^{59}Co per kilogram of metal. It is conservatively assumed that all of the cobalt is ^{59}Co . When ^{59}Co absorbs a neutron, it becomes ^{60}Co .

5.2.1.1 Directly Loaded Fuel Neutron Source

As described in Section 5.2.1, a total of 1,080 neutron source terms have been calculated for each directly loaded fuel assembly. Neutron source terms have been rebinned onto the MCBEND 28 group structure, shown in Table 5.2-21. The neutron source results from actinide spontaneous fission and from (α, n) reactions with oxygen in UO_2 . The isotopes ^{242}Cm and ^{244}Cm characteristically produce all but a few percent of the spontaneous fission neutrons and (α, n) source in light water reactor fuel. The next largest contribution is from (α, n) reactions of ^{238}Pu with oxygen. The neutron spectrum from spontaneous fission is based on fission spectrum

measurements of ^{235}U and ^{252}Cf . Neutron spectra from (α, n) reactions are based on Po- α -O source measurements. These spectra are included in the ORIGEN-S nuclear data libraries of the SCALE 4.3 code package. The spectra are automatically collapsed from the energy group structure of the data library into that of the standard MCBEND 28 group structure using ORIGEN-S as part of the source term decay evaluation.

The effect of subcritical neutron multiplication is not directly computed in the MCBEND analysis conducted for directly loaded fuel, due to difficulties in adequately biasing the calculation. Instead, neutron source rates are scaled by a subcritical multiplication factor based on the system multiplication factor, k_{eff} :

$$\text{Scale Factor} = \frac{1}{1 - k_{\text{eff}}}$$

For the dry cask conditions of transport, the system k_{eff} is taken as 0.4, with a resulting scale factor of 1.67. This scale factor is input as a scaling factor on the source strength input in MCBEND.

5.2.1.2 Directly Loaded Fuel Gamma Sources

As described in Section 5.2.1, 1,080 gamma source terms have been calculated for each directly loaded fuel assembly. Gamma source terms have been rebinned onto the MCBEND 22 group structure, shown in Table 5.2-22, using ORIGEN-S as part of the source term decay evaluation. The hardware gamma spectrum for directly loaded fuel contains contributions primarily from ^{60}Co due to the activation of Type 304 stainless steel with 1.2 g/kg ^{59}Co impurity and with some minor contributions from ^{59}Ni and ^{58}Fe . The magnitude of these spectra is based on the irradiation of 1 kg of stainless steel in the in-core flux spectrum produced by the SAS2H neutronics calculation.

The activated fuel assembly hardware source terms are found by multiplying the source strength from 1 kilogram by the kilograms of steel or inconel material in the plenum, upper end fitting or lower end fitting regions, and by multiplying by a regional flux ratio. The regional flux ratio accounts for the effects of both magnitude and spectrum variation on hardware activation. These ratios are based on empirical data (Luksic). A flux ratio of 0.2 is applied to hardware regions directly adjacent to the active core region (i.e., upper and lower plenum) and a flux ratio of 0.1 is

applied to hardware regions once removed from the active core region (i.e., upper and lower end fitting region). Activated mass in each region and the corresponding flux factor are summarized in Table 5.2-4 for each array size. In the case of CE 16x16 fuel, which has a longer plenum, the upper end fitting (upper nozzle) flux factor is reduced to 0.05 (Luksic).

5.2.1.3 Directly Loaded Fuel Source Axial Profiles

The design basis axial burnup profile used in the directly loaded fuel shielding evaluations is shown in Figure 5.2-1. This profile is converted to a set of line segments suitable for input into the MCBEND shielding code. The converted profile assures that the burnup peak of the more detailed profile shown in Figure 5.2-1 is bounded while conserving the average burnup (i.e., the area under the curve is maintained as to not add source). Neutron and gamma source profiles are computed based on the relation between burnup, B, and source strength, S, in the form:

$$S = aB^b$$

where parameters a and b are determined based on fits to SAS2H computed source rates at various fuel burnups. The parameter a is simply a scaling factor and is not relevant to the analysis. For neutron sources, parameter b is 4.22. For gamma sources, the relation between burnup and source rate is linear and b is 1.0. The resulting gamma source profile is therefore identical to the burnup profile. Table 5.2-5 gives the resulting neutron and gamma source rate profiles for directly loaded fuel. The relative source strength in each axial interval is shown, and these values are used directly in the MCBEND source strength description by defining an axial source mesh within the fuel region at the indicated elevations for each fuel type. A plot of the axial source profiles is shown in Figure 5.2-2.

5.2.2 Yankee Class Fuel and GTCC Waste Source Specification

The canistered fuel design basis source terms are based on the CE 16 x 16 Yankee Class fuel assembly with a burnup of 36,000 MWD/MTU and 8.1-year cooling time. An enrichment of 3.7 wt % ^{235}U is selected to maximize the neutron source for this type of fuel. Dose rates associated with the Yankee Class Westinghouse, United Nuclear, and CE (3.5-wt % ^{235}U) fuel types at 32,000 MWD/MTU are bounded by the canister fuel design basis for cooling times of 22, 11 and 7 years, respectively. Exxon fuel at 36,000 MWD/MTU with steel or Zircaloy fuel hardware is bounded by the canister fuel design basis for cooling times of 16 and 10 years, respectively. The 10-year cool time for Exxon fuel assemblies with Zircaloy fuel region hardware includes the activation of four hollow Zircaloy replacement rods with stainless steel slugs. Multiple

inert/dummy rod configurations were employed in CE and Exxon fuel assemblies, while only the stainless steel slugs are considered in the shielding evaluation based on their contribution to the activated hardware source term. Also of note, the cool time of 22 years for Westinghouse fuel assemblies is based on the activation of 120 kg of fuel region hardware.

Neutron and gamma source terms for the canistered design basis fuel are calculated with the SAS2H code sequence of the SCALE 4.3 code package for the PC. SAS2H includes an XSDRNPM neutronics model of the fuel assembly and ORIGEN-S fuel depletion/source term calculations. The canister fuel assembly input data for SAS2H is summarized in Table 5.2-7. Source terms are generated for both UO_2 fuel and fuel assembly hardware. The hardware activation is calculated by light element transmutation using the in-core neutron flux spectrum produced by the SAS2H neutronics model. The hardware is assumed to be Type 304 stainless steel with 1.2 g/kg of ^{59}Co impurity. The effects of axial flux spectrum and magnitude variation on hardware activation are estimated by flux ratios based on empirical data (Luksic).

5.2.2.1 Yankee Class Fuel Neutron Source

The design basis fuel neutron spectrum for canistered fuel is shown in Table 5.2-8. The neutron source results from actinide spontaneous fission and from (α, n) reactions with oxygen in UO_2 . The isotopes ^{242}Cm and ^{244}Cm characteristically produce all but a few percent of the spontaneous fission neutrons and (α, n) source in light water reactor fuel. The next largest contribution is from (α, n) reactions of ^{238}Pu with oxygen. The neutron spectra from spontaneous fission is based on fission spectrum measurements of ^{235}U and ^{252}Cf . Neutron spectra from (α, n) reactions are based on Po- α -O source measurements. These spectra are included in the ORIGEN-S nuclear data libraries of the SCALE 4.3 code package. The spectra are automatically collapsed from the energy group structure of the data library into that of the SCALE 27 group neutron cross section library.

5.2.2.2 Yankee Class Fuel and Yankee GTCC Waste Gamma Sources

The design basis Yankee Class fuel gamma spectrum for canistered fuel is shown in Table 5.2-9. Fuel gamma radiation sources consist primarily of decay gammas from fission products. Actinides also emit a significant amount of gamma radiation. The gamma source strength depends on the irradiation period and the cooling time after discharge from the reactor core.

An additional source of gamma radiation is from ^{59}Co activation in the fuel hardware materials. The design basis fuel hardware gamma spectrum for canistered fuel is shown in Table 5.2-10.

The gamma spectrum for the decay of ^{60}Co in the activated hardware was calculated using ORIGEN-S. The total source in each hardware region depends on the flux used to irradiate the region and the mass of material in that region. The default gamma energy group spectrum of the hardware gamma sources were rebinned to the SCALE-4.0 18-group structure using ORIGEN-S. This method regroups the source based on the actual energy spectrum of each specific nuclide, yielding more accurate results than those achieved by simply multiplying the individual energy group source strength by the ratio of the old to new mean energies of each respective group.

The hardware gamma spectra contains contributions primarily from ^{60}Co due to the activation of Type 304 stainless steel with 1.2 g/kg ^{59}Co impurity and with some minor contributions from ^{59}Ni and ^{58}Fe . The magnitude of these spectra is based on the irradiation of 1 kg of stainless steel in the incore flux spectrum produced by the SAS2H neutronics calculation. This activated hardware spectra is used for the GTCC waste spectra, but the magnitude is scaled up from 103 curies of ^{60}Co in the 1 kg of activated hardware to 1.25×10^5 curies ^{60}Co in the GTCC waste. This assumed source term bounds the source term for the fuel assembly cage and depleted source vane material placed in an interior loading position of the GTCC basket.

The activated fuel assembly hardware source terms are found by multiplying the source strength from 1 kilogram by the kilograms of steel or inconel material in the plenum, upper end fitting or lower end fitting regions, and by multiplying by a regional flux ratio. The regional flux ratio accounts for the effects of both magnitude and spectrum variation on hardware activation. These ratios are based on empirical data (Luksic). A flux ratio of 0.2 is applied to hardware regions directly adjacent to the active core region (i.e., upper and lower plenum) and a flux ratio of 0.1 is applied to hardware regions once removed from the active core region (i.e., upper and lower end fitting region).

5.2.2.3 Yankee Class Fuel Source Axial Profiles

The design basis Yankee Class fuel axial burnup profile used in the canister fuel shielding evaluation is shown in Figure 5.2-3. This is based on core calculations of Yankee Class fuel in the range of 30,000 to 36,000 MWD/MTU of burnup. This burnup profile has a peaking factor of 1.15. Thus, a peaking factor of 1.15 is applied to the radial midplane gamma dose rates and a peaking factor of $(1.15)^{4.2} = 1.80$ is applied to the radial midplane neutron dose rates reported in Tables 5.1-6 and 5.1-7.

The design basis gamma source profile for Yankee GTCC waste (primarily activated stainless steel core baffle) is shown in Figure 5.2-4. A GTCC gamma source peaking factor of 1.23 for GTCC waste is determined from actual dose rate measurements of the GTCC waste containers holding core baffle material. This peaking factor is due to the activation of the core baffle from 30 years of neutron flux exposure. This neutron flux exposure produces an activation profile similar to the chopped cosine axial shape of the neutron flux during reactor operation. The GTCC source term includes an estimated contribution from crud (as surface contamination).

5.2.3 Connecticut Yankee Fuel and GTCC Waste Source Specification

The NAC-STC loaded with the CY-MPC system is designed to safely transport Connecticut Yankee (CY) fuel assemblies, non-fuel hardware, and GTCC waste. The spent fuel inventory consists of both stainless steel and Zircaloy clad fuel assemblies. Due to the activation of cobalt impurities present in the stainless steel cladding, these two fuel types are considered separately in the analysis. Based on the fuel inventory, the limiting combination of burnup and cool time for the two fuel types is:

Fuel Type	Maximum Burnup [MWD/MTU]	Minimum Cool Time [years]
Stainless steel clad	38,000	10
Zircaloy clad	43,000	10

In some cases, additional activated non-fuel hardware, such as Reactor Control Cluster Assemblies and Flow Mixers, will be inserted in CY fuel assemblies. The fuel assemblies with inserted components are subject to certain additional constraints on either maximum burnup or minimum cool time. However, no credit for the reduced source terms of these fuel assemblies is taken here. The non-fuel hardware activation analysis is performed using conservative values for cumulative lifetime exposure. Each Reactor Control Cluster Assembly is assumed to have been positioned in the reactor during operation so as to achieve the maximum possible activation of material throughout its lifetime. Additionally, each flow mixer is assumed to have been in place in the top nozzle of a fuel assembly for every CY plant cycle.

The CY-MPC may also contain up to two assemblies with a maximum of two irradiated stainless steel filler rods per assembly. The mass of irradiated steel is calculated based on the maximum CY fuel rod diameter of 1.076 cm, a rod length of 321.87 cm, and the stainless steel density; the calculated mass is 2.323 kg. Conservatively, the entire mass is applied over the active fuel height.

The design basis CY GTCC waste inventory is analyzed by assuming a fixed mass loading in the CY-MPC GTCC basket and applying the highest activity of any of the GTCC sources to the mass loading. The design basis GTCC source is the core baffle, which has a modeled total activity of 3.71×10^5 curies at 10 years from core shutdown (^{60}Co activity of 1.96×10^5 curies). The GTCC source is loaded into the four center tubes with a weight of 1300 lbs and the 20 peripheral waste tubes with a weight of 800 lbs for a total assumed weight of 21,200 pounds.

An evaluation of the Connecticut Yankee spent fuel inventory establishes the Westinghouse 15 x 15 fuel assembly as the limiting fuel type based on initial mass loading of uranium. The bounding stainless steel clad fuel in the Connecticut Yankee inventory has a maximum burnup of 38,000 MWD/MTU, minimum initial enrichment of 3.65 wt. % ^{235}U , and a minimum cooling time of 10 years. Some stainless steel clad fuel assemblies in the Connecticut Yankee inventory have initial enrichments as low as 3.00 wt % ^{235}U ; however, stainless steel clad assemblies with initial enrichments lower than 3.65 wt % ^{235}U also have lower maximum burnups as shown below.

For Zircaloy clad fuels, the limiting fuel description is based on a maximum burnup of 43,000 MWD/MTU, minimum initial enrichment of 3.59 wt % ^{235}U , and a minimum cooling time of 10 years. Some Zircaloy clad fuel in the Connecticut Yankee inventory have lower initial enrichments (as low as 2.95 wt % ^{235}U); however, source terms produced by the combination of burnup and initial enrichment for the limiting Zircaloy clad fuel assemblies bound those of Zircaloy clad fuel assemblies at lower initial enrichments.

The following provides the acceptable combinations of maximum burnup and minimum initial enrichment that are bounded by the design basis stainless steel and Zircaloy clad fuel assemblies utilized in the dose rate analyses:

Stainless Steel Clad Fuel			Zircaloy Clad Fuel		
Enrichment	Maximum Burnup	Minimum Cool Time	Enrichment	Maximum Burnup	Minimum Cool Time
[wt % ^{235}U]	[MWD/MTU]	[years]	[wt % ^{235}U]	[MWD/MTU]	[years]
$e \geq 3.65$	38,000	10	$e \geq 3.59$	43,000	10
$3.23 \leq e < 3.96$	34,000	10	$3.40 \leq e < 3.59$	40,000	10
$3.00 \leq e < 3.23$	30,000	10	$2.95 \leq e < 3.40$	30,000	10

The SAS2H code sequence (Herman) is used to generate source terms. This code sequence is part of the SCALE 4.3 code package for the PC (ORNL). SAS2H includes an XSDRNPM (Greene) neutronics model of the fuel assembly and ORIGEN-S (Herman) fuel depletion/source term calculations. Reactor operating conditions assumed for the analysis are shown in Table 5.2-11. The SAS2H-generated source spectra are rebinned onto the standard 28 group neutron and 22 group gamma scheme used in MCBEND as shown in Table 5.2-21 and Table 5.2-22, respectively. Source terms are generated for the fuel and fuel assembly hardware. The hardware activation is calculated by light element transmutation using the in-core neutron flux spectrum produced by the SAS2H neutronics model.

The Connecticut Yankee design basis fuel source terms are presented in Tables 5.2-12 and 5.2-13. The activated hardware source term is provided on a per unit mass basis. Source strengths are defined for five source regions: active fuel, upper end fitting, upper plenum, lower end fitting and lower plenum. The fuel assembly length, active fuel region length and fuel assembly hardware lengths are shown for the design basis fuel assemblies in Figure 5.1-5.

5.2.3.1 Connecticut Yankee Fuel Gamma Source

The design basis fuel and hardware gamma source spectra are shown in Tables 5.2-11 and 5.2-13. The fuel gamma source contains contributions from both fission products and actinides. The spectra are presented in the standard 22 group structure used by MCBEND. The hardware gamma spectra contains contributions primarily from ^{60}Co , due to the activation of Type 304 stainless steel with either 0.5 g/kg for stainless steel clad fuel (Table 5.2-12) or 1.2 g/kg for Zircaloy clad fuel (Table 5.2-13) ^{59}Co impurity and with some minor contributions from ^{59}Ni and ^{58}Fe . The magnitude of this spectra is based on the irradiation of 1 kg of stainless steel in the in-core flux spectrum produced by the SAS2H neutronics calculation.

The activated fuel assembly hardware source strength for a given source region is determined as the product of 1) the hardware source strength per unit mass (from Table 5.2-12 and Table 5.2-13); 2) the mass of hardware present (from Table 5.2-14); 3) a mass scale factor discussed below (Table 5.2-14); and 4) a regional flux activation ratio (Table 5.2-14). The mass scale factor simply accounts for the difference in assumed cobalt concentration in the fuel cladding for stainless steel clad fuels. For stainless steel clad fuel, the SAS2H analysis is conducted based on a cobalt concentration of 0.5 g/kg. The hardware source terms in Table 5.2-12 reflect this assumed cobalt loading. However, this cobalt concentration is used only for the fuel cladding and end plugs; the remaining hardware source regions are modeled at the standard 1.2 g/kg

cobalt concentration. Hence, a mass scale factor of $1.2/0.5 = 2.4$ is applied to these source regions for the stainless steel clad fuels only. The Zircaloy clad fuel source terms are generated based on an assumed cobalt concentration of 1.2 g/kg for all fuel hardware, except the Zircaloy cladding material, so no adjustment is required and the mass factor for Zircaloy clad fuels is unity. The regional flux activation ratio accounts for the effects of both magnitude and spectrum variation on hardware activation. These ratios are based on empirical data (Luksic). A flux ratio of 0.2 is applied to hardware regions directly adjacent to the active core region, i.e., upper and lower plenum. A flux ratio of 0.1 is applied to hardware regions once removed from the active core region, i.e., upper and lower end fitting region.

5.2.3.2 Connecticut Yankee Fuel Neutron Source

The neutron source results from actinide spontaneous fission and from (α, n) reactions with the oxygen in UO_2 . The isotopes ^{242}Cm and ^{244}Cm characteristically produce all but a few percent of the spontaneous fission neutrons and (α, n) source in light water reactor fuel. The next largest contribution is from (α, n) reactions from ^{238}Pu . The neutron spectra from spontaneous fission is based on fission spectrum measurements of ^{235}U and ^{252}Cf . Neutron spectra from (α, n) reactions is based on Po- α -O source measurements. These spectra are included in the ORIGEN-S nuclear data libraries of the SCALE 4.3 code package. The spectra is automatically collapsed from the energy group structure of the data library into that of the MCBEND 28 group neutron cross section structure.

The effect of subcritical neutron multiplication is not directly computed in the MCBEND analysis conducted here, due to difficulties in adequately biasing the calculation. Instead, neutron source rates are scaled by a subcritical multiplication factor based on the system multiplication factor, k_{eff} :

$$\text{Scale Factor} = \frac{1}{1 - k_{\text{eff}}}$$

For the dry cask conditions of transport, the system k_{eff} is taken as 0.4, with a resulting scale factor of 1.67.

5.2.3.3 Connecticut Yankee Non-Fuel Hardware Source

Activated non-fuel hardware suitable for in situ storage in the CY-MPC includes Reactor Control Cluster Assemblies and Flow Mixer hardware. There are no source term constraints on the selection of the assembly holding these components. In the case of Flow Mixer hardware, a constraint on the permissible number and location of containing assemblies within the basket is imposed. No more than eight assemblies containing Flow Mixers can be loaded into a single canister, and the assemblies must be loaded into fuel assemblies in the center-most basket positions 7, 8, 12, 13, 14, 15, 19 and 20 as shown in Figure 6.3-3.

SAS2H models are developed which reproduce the irradiation history of the Reactor Control Cluster Assemblies (RCCA) and Flow Mixers in the Connecticut Yankee spent fuel pool inventory. The cycle burnup history for the Connecticut Yankee reactor is shown in Table 5.2-15. The Flow Mixers are conservatively assumed to be present in every cycle of operation. The Reactor Control Cluster Assemblies are classified into two groups and analyzed separately. The first set of RCCAs was irradiated during operating cycles 1 through 14. The second, and more limiting group of RCCAs, was irradiated during operating cycles 15 through 19 and are assumed to have been partially inserted into the active core during operation. That is, the bottom-most 9 inches of the RCCA are exposed to 60% of the full power flux for the entire irradiation history. The source term analysis for the RCCAs considers activation of both Inconel and Ag-In-Cd material. The resulting source spectra are presented in Table 5.2-16 on a per unit mass basis. The modeled mass in each source region is shown in Table 5.2-17 for both Reactor Control Cluster Assemblies and Flow Mixers.

In general, the dose rates resulting from activation of the non-fuel hardware material are relatively small as compared to those from fuel sources. Results are presented in Section 5.4.

5.2.3.4 Connecticut Yankee Fuel Source Axial Profile

An enveloping axial burnup shape for three-dimensional shielding and thermal evaluations is created based on measured burnup profile data for PWR fuel. Neutron and gamma source profiles are computed based on an assumed relation between burnup, B , and source strength, S , in the form:

$$S = aB^b$$

where parameters a and b are determined based on fits to SAS2H computed source rates at various fuel burnups. The parameter a is simply a scaling factor and is not relevant to the analysis. For neutron sources parameter b is 4.22. For gamma sources, the relation between burnup and source rate is linear and b is 1.0. Table 5.2-18 gives the resulting source rate profiles for stainless steel clad and Zircaloy clad fuel types. The relative source strength in each axial interval is shown, and these values are used directly in the MCBEND source strength description by defining an axial source mesh within the fuel region at the indicated elevations for each fuel type. A plot of the axial source profiles is shown in Figure 5.2-5. The maximum peaking factors produced by these profiles is lower than the peaking factors reported for the Yankee Class fuel assemblies in Section 5.2.2.3 due to the higher burnup of the CY design basis fuel.

5.2.3.5 Connecticut Yankee GTCC Waste Source

The CY GTCC waste consists of stainless steel sections of the Connecticut Yankee reactor core baffle, core barrel, core support plate and miscellaneous related stainless steel hardware associated with the core components. As described in this section, the CY GTCC source term is based on a characterization of the CY GTCC material performed in September 2000.

The design basis CY GTCC source spectrum and isotopic content are shown in Tables 5.2-19 and 5.2-20, respectively. The CY GTCC waste spectrum is dominated by ^{60}Co . No axial profile is applied.

The CY GTCC source term was generated in four steps. First, the waste characterization was assessed to determine the maximum source component in curies/gram. The core baffle, with a mass of 14,300 lbs, provided the maximum curie content using this measure. Second, ORIGEN-S was used to decay the core baffle radionuclide inventory to a cool time consistent with the 10 year cool time applied to Connecticut Yankee fuel. Since the waste characterization was performed in September 2000, a decay of 6 years was specified in ORIGEN-S based on the end of Cycle 19 (Table 5.2-15). This decayed curie inventory is the characterized inventory shown in Table 5.2-20. Third, the curie content of each radionuclide was scaled up by 5% and a source spectrum was calculated in ORIGEN-S using the scaled curie content. This is the "Modeled Source" inventory shown in Table 5.2-20, which is the basis for the spectrum shown in Table 5.2-18.

The fourth step is the MCBEND source term and volume fraction calculation, summarized in the table below. MCBEND requires source strength input on a volumetric basis, which necessitates

the calculation of the equivalent source volume (24 GTCC openings with width of 8.75 inches and can height of 133.50 inches). The volume fraction of steel in the openings (0.2741) is calculated based on loading 800 lbs in each of 24 openings with the calculated source volume ($4.020\text{E}+06 \text{ cm}^3$). Next, the modeled source strength is calculated based on the ORIGEN-S total source and the mass ratio of the canister contents to the ORIGEN-S input mass ($19,200/14,300=1.34$). Finally, the volumetric source ($3.4245\text{E}+09 \text{ g/sec/cm}^3$) is calculated using the modeled source strength and the source volume. Source strength is input into MCBEND by using the spectrum from Table 5.2-19 and a volumetric component ($3.3401\text{E}-07/\text{cm}^3$) of the ratio of the volumetric source strength to the ORIGEN-S source. A source representative of 1300 lbs of CY GTCC waste is modeled in the four center tubes by introducing radial scale factors in MCBEND. No credit is taken for the additional modeled material in the four center tubes in the shielding analysis. Thus, the total effective source considered is 21,200 lbs.

Parameter	Value	Units
Maximum Source Weight	19,200	lbs
Core Baffle Source Weight	14,300	lbs
Total Source Strength (based on 14,300 lbs)	$1.03\text{E}+16$	g/sec
Modeled Source Strength (based on 19,200 lbs)	$1.38\text{E}+16$	g/sec
Total Decay Heat - 14,300 lbs	$2.04\text{E}+03$	watts
Equivalent Source Area	11854.82	cm^2
Equivalent Source Length	339.090	cm
Equivalent Source Volume	$4.020\text{E}+06$	cm^3
SS304 Density	0.017424	lbs/cm^3
Weight at Full Density	$7.004\text{E}+04$	lbs
Maximum Weight (19,200 lbs) Volume Fraction	0.2741	
Maximum Volumetric Source	$3.4245\text{E}+09$	g/sec/cm^3
MCBEND Source Component	$3.3401\text{E}-07$	$/\text{cm}^3$

Figure 5.2-1 Directly Loaded Fuel Design Basis Burnup Profile

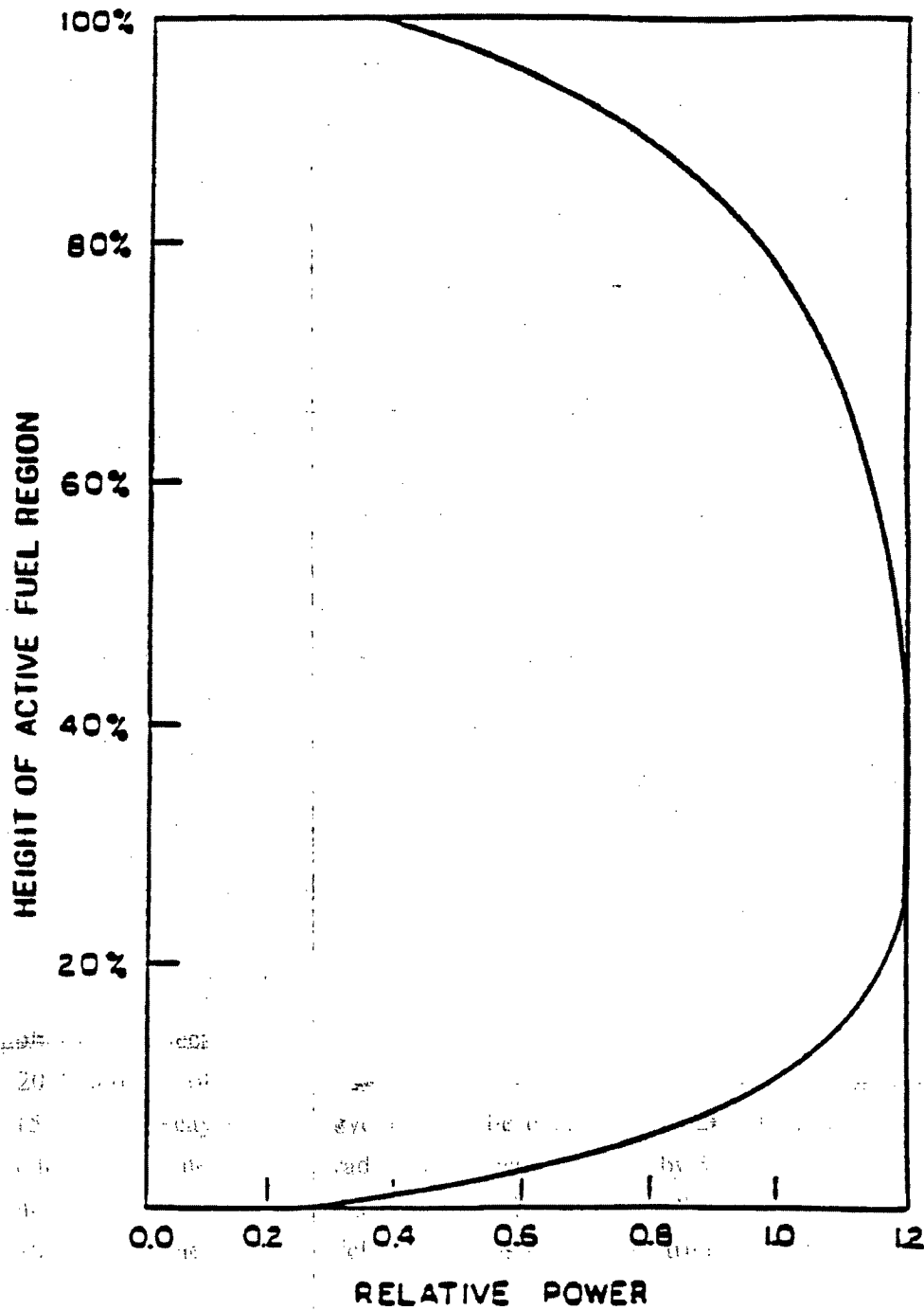


Figure 5.2-2 Directly Loaded Fuel Neutron and Gamma Source Profiles

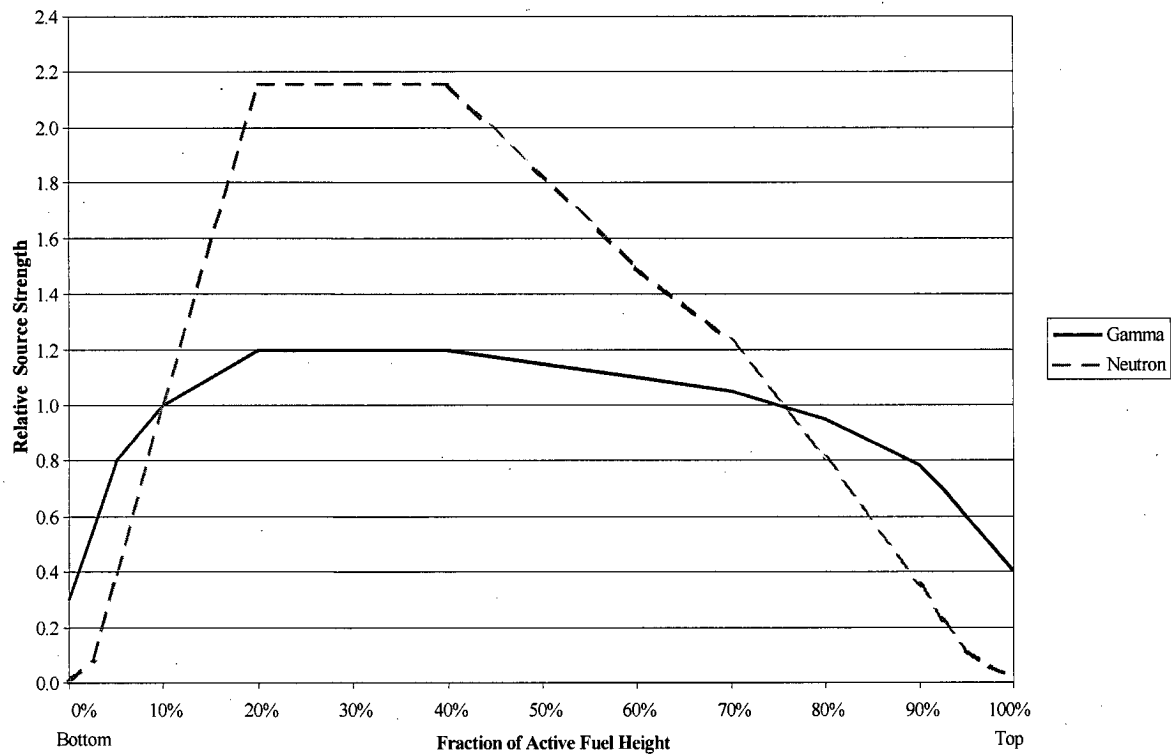


Figure 5.2-3 Design Basis Yankee Class Fuel Burnup Profile

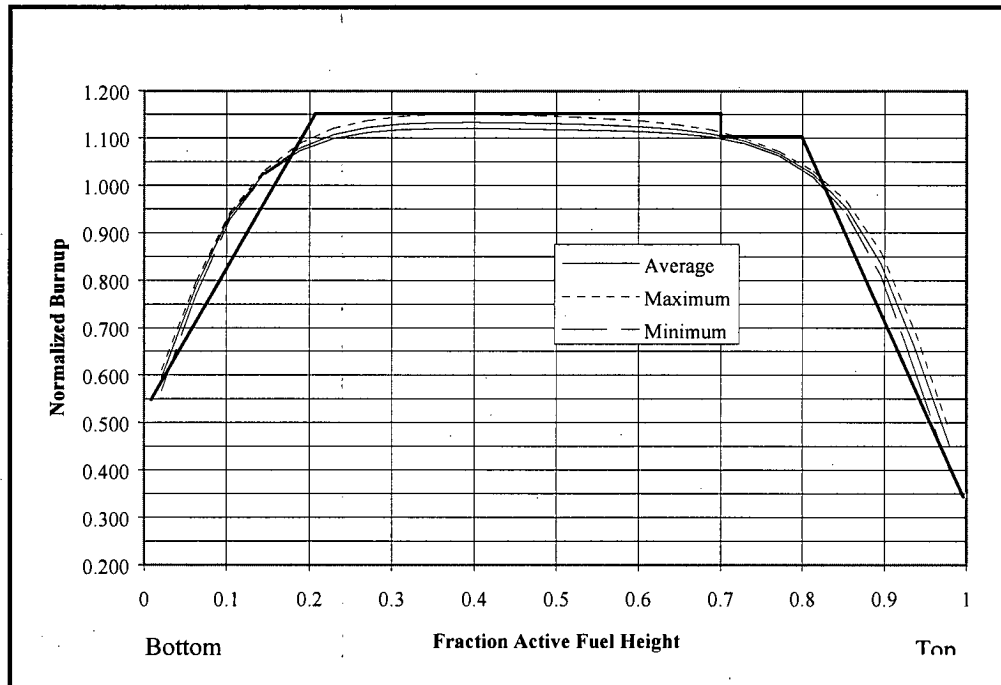


Figure 5.2-4 Yankee GTCC Waste Container Gamma Source Profile Based on Dose Rate Measurements

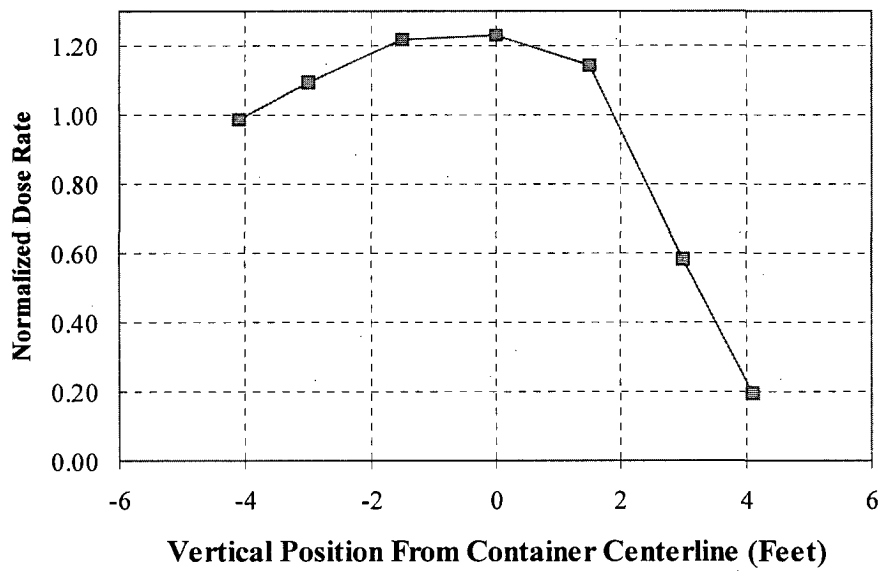


Figure 5.2-5 Connecticut Yankee Design Basis Fuel Neutron and Gamma Burnup Profiles

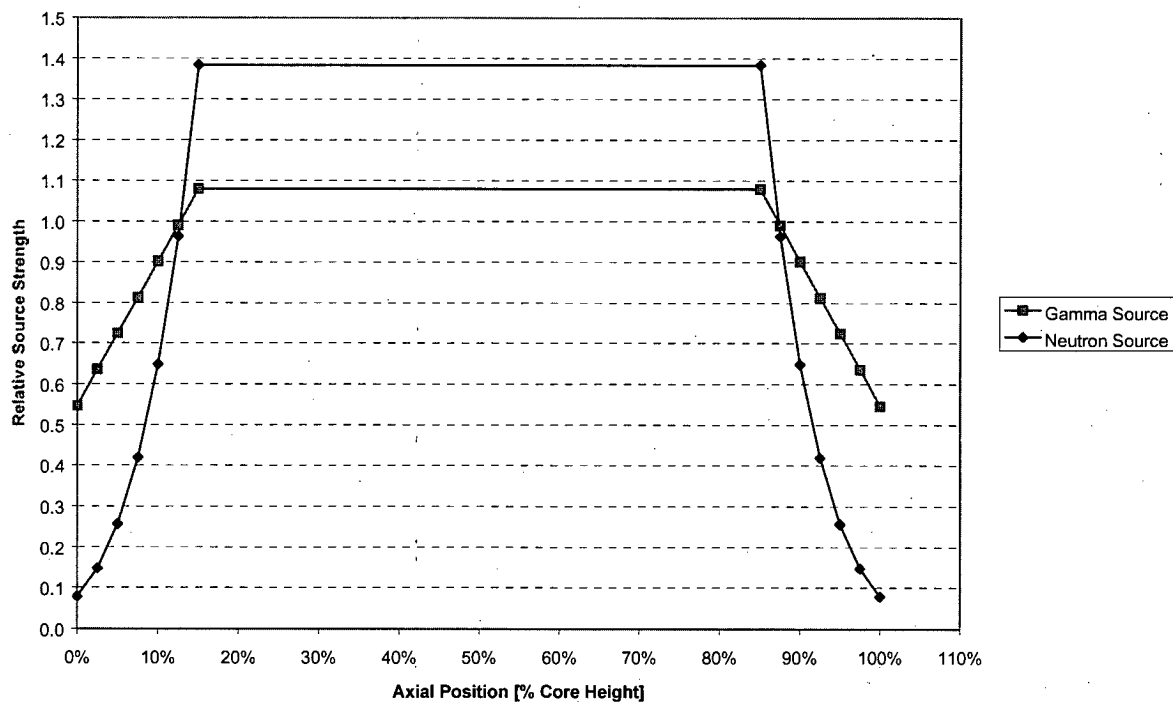


Table 5.2-1 Directly Loaded Three-Dimensional PWR Reference Fuel Assembly Descriptions

Parameter Description	Reference Fuel Assembly			
	14×14	15×15	16×16	17×17
Fuel Rod Height [inch]	152.360	152.756	146.499	152.300
Top End-Cap Height [inch]	0.685	0.685	0.750	0.685
Bottom End-Cap Height [inch]	0.685	0.685	0.891	0.895
Active Fuel Region Height [inch]	145.2	144.0	136.7	144.0
Fuel Rod Diameter [inch]	0.422	0.422	0.382	0.374
Fuel Clad Thickness [inch]	0.023	0.024	0.025	0.022
Fuel Pellet Diameter [inch]	0.367	0.367	0.325	0.323
Array	14	15	16	17
Fuel Rod Pitch [inch]	0.556	0.563	0.506	0.496
Number of Guide Tubes	16	20	4	24
Guide Tube OD [inch]	0.481	0.484	1.115	0.474
Guide Tube Thickness [inch]	0.034	0.015	0.026	0.015
Number of Instrument Tubes	1	1	1	1
Instrument Tube OD [inch]	0.481	0.484	1.115	0.474
Instrument Tube Thickness [inch]	0.034	0.015	0.026	0.015
Fuel Assembly Height [inch]	161.100	160.100	158.129	161.693
Fuel Assembly Width [inch]	7.763	8.449	8.100	8.430
Lower Nozzle Height [inch]	2.738	2.738	3.821	2.421
Upper Nozzle Height [inch]	3.500	3.480	6.821	3.480
Gap Fuel Rod to Bottom Nozzle [inch]	0.000	0.813	0.000	0.791
Gap Fuel Rod to Top Nozzle [inch]	2.502	0.313	0.988	2.701
Upper Plenum Region Height [inch]	5.790	7.386	8.158	6.720
Number of Fuel Rods	179	204	236	264
Calculated MTU [MTU]	0.4144	0.4671	0.4025	0.4636
Lower Nozzle Hardware Mass [kg]	7.893	5.680	5.400	6.307
In-core Hardware Mass [kg] ¹	14.880	17.450	1.360	5.440
Upper Plenum Hardware Mass [kg]	8.050	4.120	10.700	5.410
Upper Nozzle Hardware Mass [kg]	9.890	11.840	9.500	7.850

1. Reference literature indicates the use of stainless steel guide/instrument tubes and/or steel grids in early generations of 14×14 and 15×15 assemblies, with later generations switching to Zircaloy components. To provide a bounding fuel description, the higher steel mass present in the early generations of these fuels was employed in the source term and shielding models.

Table 5.2-2 PWR Fuel Reactor Operating Conditions for Directly Loaded Fuel

Reference Assembly Parameter	14×14	15×15	16×16	17×17
Assembly Power, MW	13.72	16.33	16.62	18.55
Fuel Temperature, K	900	900	900	900
Clad Temperature, K	620	620	620	620
Moderator Temperature, K	580	580	580	580
Moderator Density, g/cc	0.725	0.725	0.725	0.725
Boron, ppm	550	550	550	550
Down Time, days	60	60	60	60

Table 5.2-3 PWR Cycle Length Calculation for Directly Loaded Fuel Source Terms

Reference Fuel Type	Burnup [MWD/MTU]	% TD	Pellet OD [cm]	Active Length [cm]	Number of Rods	Volume [cm ³]	MTU [MTU]	Assy Power [MW]	Number of Cycles	Cycle Length [days]
14x14	30000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	2	453.14
	35000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	352.45
	40000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	402.80
	45000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	453.14
	50000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	503.49
	55000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	553.84
	60000	0.95	0.9332	368.808	179	4.52E+04	0.4144	13.72	3	604.19
15x15	30000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	2	429.15
	35000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	333.78
	40000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	381.46
	45000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	429.15
	50000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	476.83
	55000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	524.51
	60000	0.95	0.9319	365.76	204	5.09E+04	0.4671	16.33	3	572.20
16x16	30000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	2	363.26
	35000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	282.53
	40000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	322.90
	45000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	363.26
	50000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	403.62
	55000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	443.98
	60000	0.95	0.8255	347.218	236	4.39E+04	0.4025	16.62	3	484.34
17x17	30000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	2	374.82
	35000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	291.53
	40000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	333.18
	45000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	374.82
	50000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	416.47
	55000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	458.12
	60000	0.943	0.8192	365.76	264	5.09E+04	0.4636	18.55	3	499.76

Table 5.2-4 Directly Loaded PWR Fuel Assembly Hardware Mass and Activation Scale Factors by Source Region

Reference Fuel Type	Region	Activated Mass [kg/assy]	Flux Factor
14x14	Lower Nozzle	7.89	0.20
	Fuel	14.88	1.00
	Upper Plenum	8.05	0.20
	Upper Nozzle	9.89	0.10
15x15	Lower Nozzle	5.68	0.20
	Fuel	17.45	1.00
	Upper Plenum	4.12	0.20
	Upper Nozzle	11.84	0.10
16x16	Lower Nozzle	5.40	0.20
	Fuel	1.36	1.00
	Upper Plenum	10.70	0.20
	Upper Nozzle	9.50	0.05
17x17	Lower Nozzle	6.31	0.20
	Fuel	5.44	1.00
	Upper Plenum	5.41	0.20
	Upper Nozzle	7.85	0.10

Table 5.2-5 Directly Loaded Fuel Axial Gamma and Neutron Source Profiles

% Core Height	Burnup Profile	Gamma Interval	Neutron Interval
0.0	0.30	--	--
2.5	0.55	4.250E-01	4.322E-02
5.0	0.80	6.750E-01	2.351E-01
10.0	1.00	9.000E-01	6.950E-01
20.0	1.20	1.100E+00	1.579E+00
40.0	1.20	1.200E+00	2.158E+00
60.0	1.10	1.150E+00	1.827E+00
70.0	1.05	1.075E+00	1.362E+00
80.0	0.95	1.000E+00	1.017E+00
90.0	0.78	8.650E-01	5.779E-01
92.5	0.70	7.400E-01	2.862E-01
95.0	0.60	6.500E-01	1.689E-01
97.5	0.50	5.500E-01	8.474E-02
100.0	0.40	4.500E-01	3.729E-02

Table 5.2-6 Directly Loaded 14x14 Fuel Assembly Spectra at 40,000 MWD/MTU, 2.3 wt % ²³⁵U, 10 Years Cool Time in MCBEND Group Format

Group	Fuel Neutron [neutrons/sec/assy]	Fuel Gamma [photons/sec/assy]	Fuel Hardware [photons/sec/kg]
1	0.000E+00	0.0000E+00	0.0000E+00
2	2.224E+04	9.8892E+03	0.0000E+00
3	9.268E+04	1.9127E+05	0.0000E+00
4	3.079E+05	9.0084E+05	0.0000E+00
5	9.658E+05	4.5923E+06	0.0000E+00
6	2.593E+06	1.1443E+07	0.0000E+00
7	4.476E+06	3.8389E+08	2.6368E-15
8	1.498E+07	3.1958E+09	3.6815E+04
9	2.543E+07	4.1373E+10	2.3742E+07
10	3.434E+07	9.7271E+10	6.7752E-06
11	8.165E+07	2.9870E+12	6.3110E+00
12	1.275E+08	4.4115E+13	2.2492E+12
13	3.319E+07	2.0530E+13	2.3708E+12
14	1.151E+07	8.5789E+13	9.9650E+08
15	2.977E+02	1.4229E+15	4.1857E+06
16	0.000E+00	1.3765E+14	1.2053E+07
17	0.000E+00	2.5948E+13	1.9070E+08
18	0.000E+00	4.3836E+13	1.4535E+08
19	0.000E+00	1.6029E+14	2.9272E+09
20	0.000E+00	1.9875E+14	1.2134E+10
21	0.000E+00	4.8210E+14	3.4773E+10
22	0.000E+00	3.2604E+14	4.1497E+10
23	0.000E+00	--	--
24	0.000E+00	--	--
25	0.000E+00	--	--
26	0.000E+00	--	--
27	0.000E+00	--	--
28	0.000E+00	--	--
Total	3.370E+08	2.9511E+15	4.7127E+12

Table 5.2-7 Design Basis Yankee Class Fuel Input Parameters for SAS2H

Parameter	Value
Basket Configuration	Canistered
Fuel assembly type	CE 16x16 Yankee Class
Weight of U, kg/assembly	239.4
In core grids, kg/assembly	2.36 (4 Zircaloy)
Plenum spring, kg/assembly	0.762
Grids in plenum springs, kg/assembly	0.590 (Zircaloy)
Upper end fittings, kg/assembly	5.5
Lower end fittings, kg/assembly	5.2
Lower Plenum Hardware, kg/assembly	1.73
Fuel enrichment, wt.% ²³⁵ U	3.7 ¹
Fuel burnup, MWD/MTU	36,000
Cooling time	8
Burnup cycle, power cycles, down cycles	2 cycles of 496 days 1 of 60 days
Burnup, MWD/assembly	8,618
Irradiation power, MW	8.486
⁵⁹ Co concentration in steel hardware, g/kg	1.2
Irradiation flux, grid spacers in core region	100%
grid spacers in plenum region	20%
upper plenum springs	20%
upper end-fittings	10%
lower end-fittings	10%
lower plenum hardware	20%
Fuel temperature, K	787
Clad temperature, K	600
Coolant temperature, K	551
Boron content in coolant, ppm (by weight)	800

¹ The analyzed minimum enrichment for CE fuel is 3.66 wt % ²³⁵U for fuel with a maximum burnup of 36,000 MWD/MTU. This reduction in minimum enrichment does not significantly affect calculated dose rates. (See Section 5.4.1.2.)

Table 5.2-8 Design Basis Yankee Class Fuel Neutron Source Spectra at 36,000 MWD/MTU and 8 Years Cooling

GROUP	E _{HI} (MeV)	E _{LOW} (MeV)	Neutrons/Sec-Assembly
1	2.00E+01	6.43E+00	1.2290E+06
2	6.43E+00	3.00E+00	1.4080E+07
3	3.00E+00	1.85E+00	1.5760E+07
4	1.85E+00	1.40E+00	8.7930E+06
5	1.40E+00	9.00E-01	1.1840E+07
6	9.00E-01	4.00E-01	1.2870E+07
7	4.00E-01	1.00E-01	2.5190E+06
8	1.00E-01	1.70E-02	0.0000E+00
9	1.70E-02	3.00E-03	0.0000E+00
10	3.00E-03	5.50E-04	0.0000E+00
11	5.50E-04	1.00E-04	0.0000E+00
12	1.00E-04	3.00E-05	0.0000E+00
13	3.00E-05	1.00E-05	0.0000E+00
14	1.00E-05	3.05E-06	0.0000E+00
15	3.05E-06	1.77E-06	0.0000E+00
16	1.77E-06	1.30E-06	0.0000E+00
17	1.30E-06	1.13E-06	0.0000E+00
18	1.13E-06	1.00E-06	0.0000E+00
19	1.00E-06	8.00E-07	0.0000E+00
20	8.00E-07	4.00E-07	0.0000E+00
21	4.00E-07	3.25E-07	0.0000E+00
22	3.25E-07	2.25E-07	0.0000E+00
23	2.25E-07	1.00E-07	0.0000E+00
24	1.00E-07	5.00E-08	0.0000E+00
25	5.00E-08	3.00E-08	0.0000E+00
26	3.00E-08	1.00E-08	0.0000E+00
27	1.00E-08	1.00E-11	0.0000E+00
TOTAL			6.7090E+07

Table 5.2-9 Design Basis Yankee Class Fuel Gamma Source Spectra at 36,000 MWD/MTU and 8 Years Cooling

GROUP	E _{HI} (MeV)	E _{LOW} (MeV)	Photons/Sec-Assembly
1	1.00E+01	8.00E+00	3.7701E+04
2	8.00E+00	6.50E+00	1.7759E+05
3	6.50E+00	5.00E+00	9.0547E+05
4	5.00E+00	4.00E+00	2.2566E+06
5	4.00E+00	3.00E+00	6.2676E+08
6	3.00E+00	2.50E+00	5.1211E+09
7	2.50E+00	2.00E+00	1.0789E+11
8	2.00E+00	1.66E+00	9.9933E+10
9	1.66E+00	1.33E+00	4.8070E+12
10	1.33E+00	1.00E+00	3.4718E+13
11	1.00E+00	8.00E-01	6.3503E+13
12	8.00E-01	6.00E-01	8.2333E+14
13	6.00E-01	4.00E-01	1.1897E+14
14	4.00E-01	3.00E-01	1.7831E+13
15	3.00E-01	2.00E-01	2.8386E+13
16	2.00E-01	1.00E-01	1.0201E+14
17	1.00E-01	5.00E-02	1.3136E+14
18	5.00E-02	1.00E-02	4.5899E+14
Total			1.7842E+15

Table 5.2-10 Design Basis Yankee Class Fuel Hardware and GTCC Waste Gamma Spectra

GROUP	E _{HI} (MeV)	E _{LOW} (MeV)	Photons/Sec-kg
1	1.00E+01	8.00E+00	0.0000E+00
2	8.00E+00	6.50E+00	0.0000E+00
3	6.50E+00	5.00E+00	0.0000E+00
4	5.00E+00	4.00E+00	0.0000E+00
5	4.00E+00	3.00E+00	1.0141E-15
6	3.00E+00	2.50E+00	3.3511E+04
7	2.50E+00	2.00E+00	2.1611E+07
8	2.00E+00	1.66E+00	9.5163E-03
9	1.66E+00	1.33E+00	9.1066E+11
10	1.33E+00	1.00E+00	3.2247E+12
11	1.00E+00	8.00E-01	4.3841E+09
12	8.00E-01	6.00E-01	3.8100E+06
13	6.00E-01	4.00E-01	1.0971E+07
14	4.00E-01	3.00E-01	1.7359E+08
15	3.00E-01	2.00E-01	1.3230E+08
16	2.00E-01	1.00E-01	2.6645E+09
17	1.00E-01	5.00E-02	1.1044E+10
18	5.00E-02	1.00E-02	5.5673E+10
TOTAL			4.2095E+12

Table 5.2-11 Connecticut Yankee Design Basis Fuel Reactor Operating Conditions

Parameter	Stainless Steel Clad	Zircaloy Clad
Assembly Power, MW	12.787	12.787
Fuel Temperature, K	900	900
Clad Temperature, K	620	620
Moderator Temperature, K	580	580
Moderator Density, g/cc	0.725	0.725
Boron, ppm	550	550
Fuel Burnup, MWD/MTU	38,000	43,000
Number of Cycles	3	3
Burnup Cycle, days	427.61	442.99
Down Time, days	60	60

Table 5.2-12 Connecticut Yankee Design Basis Stainless Steel Clad Fuel Source Term

Group	Fuel Neutron [n/sec/assy]	Fuel Gamma [g/sec/assy]	Fuel Hardware [g/sec/kg]
1	0.0000E+00	0.0000E+00	0.0000E+00
2	9.2580E+03	4.1234E+03	0.0000E+00
3	3.8570E+04	7.9755E+04	0.0000E+00
4	1.2810E+05	3.7568E+05	0.0000E+00
5	4.0200E+05	1.9154E+06	0.0000E+00
6	1.0790E+06	4.7736E+06	0.0000E+00
7	1.8630E+06	2.7498E+08	1.0419E-15
8	6.2330E+06	2.4916E+09	1.1328E+04
9	1.0640E+07	3.6651E+10	7.3057E+06
10	1.4640E+07	9.8663E+10	6.9845E-06
11	3.4830E+07	2.5367E+12	1.9419E+00
12	5.3360E+07	3.6912E+13	6.9208E+11
13	1.3830E+07	1.7642E+13	7.2951E+11
14	4.7980E+06	7.3398E+13	7.7774E+08
15	2.2220E+02	1.3845E+15	1.2880E+06
16	0.0000E+00	1.2185E+14	3.7087E+06
17	0.0000E+00	2.9407E+13	5.8681E+07
18	0.0000E+00	4.7288E+13	4.4724E+07
19	0.0000E+00	1.6792E+14	9.0073E+08
20	0.0000E+00	2.1999E+14	3.7338E+09
21	0.0000E+00	5.0942E+14	1.0749E+10
22	0.0000E+00	3.5520E+14	1.2974E+10
23	0.0000E+00		
24	0.0000E+00		
25	0.0000E+00		
26	0.0000E+00		
27	0.0000E+00		
28	0.0000E+00		
Total	1.4190E+08	2.9661E+15	1.4508E+12

Note: Source Term at 38,000 MWD/MTU at 10 years cool time.

Table 5.2-13 Connecticut Yankee Design Basis Zircaloy Clad Fuel Source Term

Group	Fuel Neutron [n/sec/assy]	Fuel Gamma [g/sec/assy]	Fuel Hardware [g/sec/kg]
1	0.0000E+00	0.0000E+00	0.0000E+00
2	1.3500E+04	6.0098E+03	0.0000E+00
3	5.6270E+04	1.1624E+05	0.0000E+00
4	1.8690E+05	5.4750E+05	0.0000E+00
5	5.3360E+07	2.7912E+06	0.0000E+00
6	1.3830E+07	6.9554E+06	0.0000E+00
7	2.7180E+06	3.0126E+08	1.6009E-15
8	9.0930E+06	2.6661E+09	3.0554E+04
9	1.5480E+07	3.7772E+10	1.9704E+07
10	2.1080E+07	1.0156E+11	6.8283E-06
11	5.0140E+07	2.7561E+12	5.2376E+00
12	7.7590E+07	4.0362E+13	1.8666E+12
13	2.0160E+07	1.9175E+13	1.9676E+12
14	6.9950E+06	8.0632E+13	8.8123E+08
15	2.4660E+02	1.4427E+15	3.4738E+06
16	0.0000E+00	1.3283E+14	1.0003E+07
17	0.0000E+00	2.9843E+13	1.5827E+08
18	0.0000E+00	4.8521E+13	1.2063E+08
19	0.0000E+00	1.7307E+14	2.4294E+09
20	0.0000E+00	2.2227E+14	1.0070E+10
21	0.0000E+00	5.2257E+14	2.8855E+10
22	0.0000E+00	3.6263E+14	3.4424E+10
23	0.0000E+00		
24	0.0000E+00		
25	0.0000E+00		
26	0.0000E+00		
27	0.0000E+00		
28	0.0000E+00		
Total	2.0570E+08	3.0776E+15	3.9112E+12

Note: Source Term at 43,000 MWD/MTU at 10 years cool time.

Table 5.2-14 Connecticut Yankee Design Basis Fuel Assembly Hardware Mass and Mass Scale Factors by Source Region

Stainless Steel Clad			
Region	Mass [kg/assembly]	Mass Factor	Activation Ratio
Lower Nozzle	8.850	2.4	0.1
Lower End Plug	2.537	1.0	0.2
Fuel	102.832	1.0	1.0
Upper Plenum	3.879	2.4	0.2
Upper End Plug	2.537	1.0	0.2
Upper Nozzle	11.240	2.4	0.1
Zircaloy Clad			
Region	Mass [kg/assembly]	Mass Factor	Activation Ratio
Lower Nozzle	5.440	1.0	0.1
Lower End Plug	0.000	1.0	0.2
Fuel	19.415	1.0	1.0
Upper Plenum	5.137	1.0	0.2
Upper End Plug	0.000	1.0	0.2
Upper Nozzle	11.840	1.0	0.1

Table 5.2-15 Connecticut Yankee Reactor Operational Cycle History

Cycle	Cycle End Date	Cycle Length (days)	Core Loading (MTU)	Avg Cycle Burnup (MWD/MTU)
1	04/17/70	838	65.9	16,965
2	04/16/71	295	65.3	7,602
3	06/10/72	382	64.7	10,201
4	07/08/73	359	64.4	9,157
5	05/17/75	519	64.6	13,152
6	05/18/76	323	64.6	8,738
7	10/15/77	454	64.6	12,283
8	01/27/79	423	64.6	11,105
9	05/03/80	418	64.7	11,044
10	09/26/81	427	64.7	11,342
11	01/22/83	437	64.7	11,081
12	08/01/84	479	64.6	12,987
13	01/04/86	422	64.5	10,987
14	07/18/87	435	64.5	10,039
15	09/02/89	526	64.5	12,982
16	10/17/91	428	64.8	10,095
17	05/15/93	426	62.5	12,035
18	01/28/95	558	59.9	13,588
19	07/22/96	461	58.5	13,844

Table 5.2-16 Connecticut Yankee Design Basis Non-Fuel Assembly Hardware Source Spectra

Group	RCCA [γ/kg/sec]	Flow Mixer	
		Upper Plenum [γ/kg/sec]	End Fitting [γ/kg/sec]
1	0.0000E+00	0.0000E+00	0.0000E+00
2	0.0000E+00	0.0000E+00	0.0000E+00
3	0.0000E+00	0.0000E+00	0.0000E+00
4	0.0000E+00	0.0000E+00	0.0000E+00
5	0.0000E+00	0.0000E+00	0.0000E+00
6	0.0000E+00	0.0000E+00	0.0000E+00
7	1.3895E-09	6.3717E-17	3.1859E-17
8	1.6188E+03	1.0762E+04	5.3810E+03
9	9.9426E+05	6.9405E+06	3.4702E+06
10	3.2170E+06	1.1419E-06	5.7094E-07
11	1.8234E+09	1.8449E+00	9.2243E-01
12	9.0559E+10	6.5749E+11	3.2874E+11
13	9.2794E+10	6.9304E+11	3.4652E+11
14	1.1895E+10	1.5598E+08	7.7991E+07
15	4.1816E+12	1.2236E+06	6.1180E+05
16	1.8817E+12	3.5233E+06	1.7617E+06
17	2.4186E+09	5.5747E+07	2.7874E+07
18	3.6601E+09	4.2489E+07	2.1244E+07
19	1.2166E+10	8.5571E+08	4.2785E+08
20	1.9099E+11	3.5472E+09	1.7736E+09
21	1.0140E+12	1.0238E+10	5.1192E+09
22	3.3857E+10	1.2439E+10	6.2194E+09
Total	7.5175E+12	1.3779E+12	6.8894E+11

Table 5.2-17 Connecticut Yankee Design Basis Non-Fuel Hardware Masses

Reactor Control Cluster Assemblies			
Source	Mass [kg]	Activation Region	Modeled Source Region
Inconel 625	1.50	Active Core (60% Flux)	Fuel (bottom 18 in)
Ag-In-Cd	5.00	Active Core (60% Flux)	Fuel (bottom 18 in)
Total	13.00	--	--
Flow Mixers			
Source	Mass [kg]	Activation and Modeled Source Region	
St. Steel	1.80	Upper Fitting	
Inconel	0.42	Upper Fitting	
Total Fitting	2.22	--	
St. Steel	2.70	Upper Plenum	
Total Plenum	2.70	--	

Table 5.2-18 CY-MPC Axial Gamma and Neutron Source Profiles – Design Basis Stainless Steel and Zircaloy Clad Fuels

% Core Height	SS Clad Elevation [cm]	Zirc Clad Elevation [cm]	Burnup Profile	Gamma Interval	Neutron Interval
0.0	14.2824	14.2824	0.5470	--	--
2.5	22.0167	21.9723	0.6358	5.914E-01	1.132E-01
5.0	29.7510	29.6621	0.7247	6.803E-01	2.024E-01
7.5	37.4853	37.3520	0.8135	7.691E-01	3.377E-01
10.0	45.2196	45.0418	0.9023	8.579E-01	5.333E-01
12.5	52.9539	52.7317	0.9912	9.468E-01	8.057E-01
15.0	60.6882	60.4215	1.0800	1.036E+00	1.173E+00
85.0	277.2486	275.7373	1.0800	1.080E+00	1.384E+00
87.5	284.9829	283.4272	0.9912	1.036E+00	1.173E+00
90.0	292.7172	291.1170	0.9023	9.468E-01	8.057E-01
92.5	300.4515	298.8069	0.8135	8.579E-01	5.333E-01
95.0	308.1858	306.4967	0.7247	7.691E-01	3.377E-01
97.5	315.9201	314.1866	0.6358	6.803E-01	2.024E-01
100.0	323.6544	321.8764	0.5470	5.914E-01	1.132E-01

Table 5.2-19 Connecticut Yankee GTCC Waste Source Term at 10 Years' Cool Time

Group	Gamma/Sec
1 - 7	0.0000E+00
8	8.0085E+07
9	5.1648E+10
10	0.0000E+00
11	1.3728E+04
12	4.8928E+15
13	5.1573E+15
14	1.5175E+12
15	4.8847E+10
16	2.6219E+10
17	4.1485E+11
18	3.1620E+11
19	6.3680E+12
20	2.6399E+13
21	7.5986E+13
22	9.1670E+13
Total ¹	1.0253E+16

1. Based on 14,300 lbs.

Table 5.2-20 Isotopic Constituents of the Connecticut Yankee GTCC Waste at 10 Years' Cool Time

Isotope	Characterized Inventory 14,300 lbs.	Modeled Source Inventory 14,300 lbs.
C 14	7.23E+01 Ci	7.59E+01 Ci
MN 54	3.15E+01 Ci	3.31E+01 Ci
FE 55	6.60E+04 Ci	6.93E+04 Ci
CO 60	1.26E+05 Ci	1.32E+05 Ci
NI 59	2.71E+02 Ci	2.85E+02 Ci
NI 63	4.61E+04 Ci	4.84E+04 Ci
NB 94	1.02E+00 Ci	1.07E+00 Ci
TC 99	1.83E-01 Ci	1.92E-01 Ci
TOTAL	2.38E+05 Ci	2.50E+05 Ci

Table 5.2-21 MCBEND Standard 28 Group Neutron Boundaries

Group	E Lower [MeV]	E Upper [MeV]	E Average [MeV]
1	1.360E+01	1.460E+01	1.410E+01
2	1.250E+01	1.360E+01	1.305E+01
3	1.125E+01	1.250E+01	1.188E+01
4	1.000E+01	1.125E+01	1.063E+01
5	8.250E+00	1.000E+01	9.125E+00
6	7.000E+00	8.250E+00	7.625E+00
7	6.070E+00	7.000E+00	6.535E+00
8	4.720E+00	6.070E+00	5.395E+00
9	3.680E+00	4.720E+00	4.200E+00
10	2.870E+00	3.680E+00	3.275E+00
11	1.740E+00	2.870E+00	2.305E+00
12	6.400E-01	1.740E+00	1.190E+00
13	3.900E-01	6.400E-01	5.150E-01
14	1.100E-01	3.900E-01	2.500E-01
15	6.740E-02	1.100E-01	8.870E-02
16	2.480E-02	6.740E-02	4.610E-02
17	9.120E-03	2.480E-02	1.696E-02
18	2.950E-03	9.120E-03	6.035E-03
19	9.610E-04	2.950E-03	1.956E-03
20	3.540E-04	9.610E-04	6.575E-04
21	1.660E-04	3.540E-04	2.600E-04
22	4.810E-05	1.660E-04	1.071E-04
23	1.600E-05	4.810E-05	3.205E-05
24	4.000E-06	1.600E-05	1.000E-05
25	1.500E-06	4.000E-06	2.750E-06
26	5.500E-07	1.500E-06	1.025E-06
27	7.090E-08	5.500E-07	3.105E-07
28	1.000E-11	7.090E-08	3.546E-08

Table 5.2-22 MCBEND Standard 22 Group Gamma Boundaries

Group	E Lower [MeV]	E Upper [MeV]	E Average [MeV]
1	1.200E+01	1.400E+01	1.300E+01
2	1.000E+01	1.200E+01	1.100E+01
3	8.000E+00	1.000E+01	9.000E+00
4	6.500E+00	8.000E+00	7.250E+00
5	5.000E+00	6.500E+00	5.750E+00
6	4.000E+00	5.000E+00	4.500E+00
7	3.000E+00	4.000E+00	3.500E+00
8	2.500E+00	3.000E+00	2.750E+00
9	2.000E+00	2.500E+00	2.250E+00
10	1.660E+00	2.000E+00	1.830E+00
11	1.440E+00	1.660E+00	1.550E+00
12	1.220E+00	1.440E+00	1.330E+00
13	1.000E+00	1.220E+00	1.110E+00
14	8.000E-01	1.000E+00	9.000E-01
15	6.000E-01	8.000E-01	7.000E-01
16	4.000E-01	6.000E-01	5.000E-01
17	3.000E-01	4.000E-01	3.500E-01
18	2.000E-01	3.000E-01	2.500E-01
19	1.000E-01	2.000E-01	1.500E-01
20	5.000E-02	1.000E-01	7.500E-02
21	2.000E-02	5.000E-02	3.500E-02
22	1.000E-02	2.000E-02	1.500E-02

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5.3 Model Specification

The radiation protection provided by the NAC-STC is in the form of solid multi-walled shielding materials which totally surround the fuel. These shielding materials include steel and lead for gamma shielding and a borated polymer (NS-4-FR) for neutron shielding. The multi-walled arrangement of steel and lead in the NAC-STC provides optimal weight for gamma attenuation. The NS-4-FR neutron shielding material has a hydrogen density close to that of water and serves to moderate fast neutrons which are then captured in the boron. Boron capture in the neutron shield minimizes the contribution of secondary capture gammas to surface dose rates.

The NAC-STC uses a multi-walled arrangement for both radial and axial shields. The arrangement of the radial gamma shielding in the cask body is a 1.5-inch thick stainless steel inner shell and a 2.65-inch thick stainless steel outer shell with a 3.70-inch lead annulus between them. The radial neutron shield is arranged around the outer steel shell with a 5.5-inch minimum, 5.925-inch maximum thickness of NS-4-FR that is covered by a 0.236-inch (6 mm) thick neutron shield shell. Variations in the neutron shield thickness result from filling the cavity formed by the multifaceted neutron shield shell and the cylindrical cask outer shell with NS-4-FR. The minimum shield thickness is obtained at the heat transfer fin location and increases towards the cask corners. As indicated in Section 5.3.2.1, an equivalent cylindrical shield has a 5.52-inch thickness (conservatively assuming that the expansion foam in the cavity retains its maximum thickness). The bottom of the cask contains a steel/NS-4-FR/steel shield arrangement with the two stainless steel components providing 11.65 inches of gamma shielding and 2 inches of NS-4-FR neutron shielding. The top of the cask has shields in the form of two closure lids. The inner lid also has a steel/NS-4-FR/steel arrangement with 6.0 inches of steel below 2 inches of NS-4-FR and 1.0 inch of steel above it. The outer lid is a 5.25-inch thick steel disk.

5.3.1 Directly Loaded Fuel Model

MCBEND three-dimensional shielding analysis allows detailed modeling of fuel assemblies, basket, and cask shield configuration, including streaming paths. For fuel assembly sources, some fuel assembly detail is homogenized in the model to simplify model input and improve computational efficiency. Thus, the three-dimensional models represent the various fuel assembly source regions as homogenized zones within the fuel tubes in the basket, but explicitly model the axial extent of the source regions. The fuel and hardware source regions of each assembly are therefore homogenized within the volumes defined by the periphery of the fuel assembly and the source region axial extents. The basket details, including support disks, heat

transfer disks, and top and bottom weldments are explicitly modeled. Cask body details include the axial extent of the cask shield as described by the License Drawings.

The geometric description of a MCBEND model is based on the combinatorial geometry system embedded in the code. In this system, bodies such as cylinders and rectangular parallelepipeds, and their logical intersections and unions, are used to describe the extent of material zones.

MCBEND employs an automated biasing technique for the Monte Carlo calculation based on a three-dimensional adjoint diffusion calculation. Mesh cells for the adjoint solution are selected based on half value thicknesses for each material.

MCBEND Monte Carlo calculations are performed for each source type present in each source region. This approach entails seven separate analyses, encompassing fuel neutron, fuel gamma, fuel n-gamma (secondary gammas arising from neutron interaction in the shield), fuel region hardware, upper plenum, and upper and lower end-fitting gamma sources. Typically, a total of 5 to 20 million histories are tracked to yield dose rate profiles for each model. These cases are analyzed for both radial and axial detector locations and for normal and hypothetical accident conditions.

5.3.1.1 Directly Loaded Assembly Model

Based on the fuel parameters provided in Table 5.2-1, homogenized treatments of fuel assembly source regions are developed. The homogenized fuel assembly is represented in the model as a stack of boxes with width equal to the fuel assembly width. The height of each box corresponds to the modeled height of the corresponding assembly region.

The active fuel region homogenizations for the four design basis assemblies are shown in Table 5.3-1 based on the detailed three-dimensional data in Table 5.2-1. Components of the fuel assembly homogenization are subdivided to account for the various area fractions present in the homogenized fuel assembly description. "Interstitial" refers to the space within the fuel assembly array defined by the lattice pitch but outside the fuel rod or guide/instrument tubes. "Inside tubes" refers to the space inside the instrument and guide tubes and "void" refers to the pellet to clad gap. All three regions are assigned a void material as part of the shielding evaluation since the cask cavity is dry during all transport conditions. Combined with the fuel rod clad, fuel material, and guide/instrument tube materials the void accounts for the total fuel region volume. The clad region is Zircaloy (density 6.55 g/cm³) for all four design basis

assemblies. The resulting regional compositions on an atom/barn-cm basis are shown in Table 5.3-2.

Fuel assembly non-fuel regions are homogenized as shown in Tables 5.3-3 and 5.3-4 for stainless steel and Zircaloy materials, respectively. Using the detailed dimensions in Table 5.2-1, the axial source regions are modeled as follows: the gap between the lower nozzle and the fuel rod (if present) and the fuel rod bottom end-cap are assigned to the lower nozzle region, the fuel rod top end-cap and gap to the top nozzle are assigned to the top plenum region, with the top nozzle region height not being modified. The only material included in the homogenized region is stainless steel for the upper end fitting and combinations of Zircaloy and stainless steel in the upper plenum and lower end fitting regions. Zircaloy in these regions is due to end caps and the portion of the fuel rod cladding in the upper plenum. Volume fractions of material are based on the modeled regional volume and the volume of stainless steel or Zircaloy present as computed from the modeled mass and density (7.92 g/cm^3 for stainless steel and 6.55 g/cm^3 for Zircaloy).

5.3.1.2 Directly Loaded Basket Model

For a given fuel type, the MCBEND description of the basket elements forms a common sub-model employed in the transport cask analysis. The key features of the model are the detailed representation of fuel tubes, basket support and heat transfer disks, and weldment structures.

5.3.1.3 Description of MCBEND NAC-STC Model

The three-dimensional model of the NAC-STC cask containing design basis fuel assemblies is based on the following features:

Normal conditions:

- Radial neutron shield and shield shell (includes heat fins in the neutron shield)
- All balsa upper and lower impact limiters (100% balsa chosen for conservatism)

Accident conditions:

- Top axial lead slump
- Bottom axial lead slump
- Radial lead slump
- Radial neutron shield and shield shell with removal of oxygen, hydrogen, and nitrogen from NS-4-FR material definition

- Loss of upper and lower impact limiters

Features common to both the normal and accident conditions models are the inner lid vent and drain ports, the inner lid neutron shield, the bottom forging neutron shield, the annular expansion foam region below the neutron shield (modeled conservatively as void), radial neutron shield heat fins, and the lower rotating trunnions.

Detailed model parameters used in creating the three-dimensional model are taken directly from the License Drawings. Elevations associated with the transport cask three-dimensional features are established with respect to the center bottom of the NAC-STC cavity for the MCBEND combinatorial model. The three-dimensional NAC-STC models are shown in Figures 5.3-1 through 5.3-3.

5.3.1.4 Directly Loaded Configuration Shield Regional Densities

Based on the homogenization described in Section 5.3.1.1, the resulting active fuel regional densities are shown in Table 5.3-2 for the design basis directly loaded fuel assemblies. Material compositions for remaining structural and shield materials are shown in Table 5.3-5. Compositions for fuel assembly non-fuel regions are equivalent to the stainless steel and Zircaloy compositions in Table 5.3-5 scaled by the material volume fractions shown in Tables 5.3-3 and 5.3-4.

5.3.2 Yankee-MPC Fuel and GTCC Waste Model Specifications

This section provides the radial and axial shielding models used for the Yankee-MPC canistered fuel and GTCC waste.

5.3.2.1 Yankee-MPC Fuel and GTCC Waste Radial Shielding Models

One-dimensional cylindrical SAS1 models are used to evaluate radial midplane dose rates for the NAC-STC containing Yankee-MPC design basis canistered fuel and GTCC waste. In both cases, the source region is transformed into an equivalent cylindrical volume. In the case of the canister fuel region, this volume is based on the periphery of the fuel basket tubes and has an equivalent radius of 30.63 inches (77.80 cm). The fuel assembly source regions are homogenized into the volumes defined by the fuel/basket equivalent radius and the fuel regional elevations defined in Figure 5.1-3. Since the canister basket contains an explicit heat transfer

region with aluminum heat transfer disks, an additional middle fuel region is defined with this material for the radial midplane evaluation. The remaining cask body regions are modeled using the exact dimensions of the cask, except for the radial neutron shield. Its thickness varies as a result of its polygon shape. An equivalent thickness of 5.52 inches (14.02 cm) is modeled to conserve the actual neutron shield volume (see Figure 5.3-4). To account for axial leakage, an axial buckling equivalent to the active fuel height is applied. In the accident situation, the neutron shield material, NS-4-FR, is voided. An axial peaking factor of 1.15 and 1.80 is applied to the radial midplane gamma and neutron results, respectively, to account for the axial burnup profile as described in Section 5.2.2. Radial models are also constructed for the Westinghouse, United Nuclear and Exxon fuel types to determine minimum cool time based on shielding constraints. By performing shielding analysis rather than source term magnitude comparisons, spectrum differences are taken into account. Shielding analysis of the Exxon assembly at 10 years' cooling is not required, since its neutron and gamma source is lower in each energy group and its mass (and therefore its self-shielding) is identical to the design basis assembly.

In the case of the GTCC waste, the volume is based on the interior periphery of the GTCC basket support wall and has an equivalent radius of 23.47 inches (59.61 cm). The GTCC source region is homogenized into the volume defined by the GTCC basket interior periphery and the container height of 98.25 inches (249.56 cm [See Figure 5.1-4]). This gives a GTCC source volume of 170,023 inches³ (2.786×10^6 cm³). The GTCC basket support wall is also cylindrical with a 2.5-inch (6.35-cm) thickness. The remaining cask body regions are modeled using the exact dimensions of the cask, except for the radial neutron shield that is again modeled with an equivalent thickness of 5.52 inches (14.02 cm [See Figure 5.3-7]). To account for axial leakage, an axial buckling equivalent to the container height of 98.35 inches is applied. In the accident situation, the neutron shield material, NS-4-FR, is voided. An axial peaking factor of 1.23 is applied to the radial midplane results to account for gamma source peaking as described in Section 5.2.2.3.

5.3.2.2 Yankee-MPC Fuel and GTCC Waste Axial Shielding Models

One-dimensional slab SAS1 models are used to evaluate top and bottom dose rates for the NAC-STC containing design basis Yankee-MPC fuel and GTCC waste. The top axial model

begins at either the active fuel or homogenized GTCC waste canister source midplane and proceeds along the cask centerline to the surface of the top impact limiter. Similarly, the NAC-STC bottom axial model begins at either the canistered fuel or homogenized GTCC waste canister center and ends at the bottom impact limiter. See Figures 5.3-5 and 5.3-6 for the canister fuel axial shielding models and Figures 5.3-8 and 5.3-9 for the canister GTCC waste axial shielding models. To account for transverse radial leakage, radial bucklings equal to the equivalent diameter of the source regions are applied. In the accident situation, the impact limiters are lost, the canister is positioned on either the top or bottom cavity surface, and the NS-4-FR neutron shield material, including that in the bottom forging, is considered to be lost.

5.3.2.3 Cask Regional Material Compositions - Yankee Class Fuel and GTCC Waste

The densities of the materials used in the shielding evaluations for canister fuel and GTCC waste are calculated using the effective fuel radius and source regional elevation. See Figure 5.1-3 for the design basis canistered fuel source zones and elevations. In the case of the canistered Yankee Class fuel, the homogenized source regions include a top fuel, middle fuel (heat transfer zone), bottom fuel, top plenum, bottom plenum, and the top and bottom end-fittings. The structural and heat transfer disks exterior to the fuel/basket region are also homogenized in the one-dimensional radial models. Similarly, the GTCC waste density is based on homogenizing the mass of GTCC waste into the volume defined by the equivalent radius and the height of the container. The homogenized densities and nuclide concentrations are shown in Table 5.3-6.

For the damaged fuel evaluation, dose rates are estimated using the amount of UO_2 that could disperse into the active fuel void space. Based on the homogenized density of UO_2 fuel of 2.2769 g/cm^3 (Table 5.3-6), additional source could be concentrated into the fuel region equivalent to the density ratio of $10.412/2.2769$, or 4.6.

5.3.3 CY-MPC Fuel and GTCC Waste Model Specifications

MCBEND three-dimensional shielding analysis allows detailed modeling of radiation source (either fuel assemblies or GTCC waste), basket, and cask shield configuration, including streaming paths. For fuel assembly sources, some fuel assembly detail is homogenized in the model to simplify model input and improve computational efficiency. Thus, the three-dimensional models represent the various fuel assembly source regions as homogenized zones within the fuel tubes in the basket, but explicitly model the axial extent of the source regions. For the GTCC waste source, the waste is homogenized using the mass of waste to be loaded in

conjunction with the available volume in the GTCC waste tubes. The basket details, including support disks, heat transfer disks, and top and bottom weldments are explicitly modeled.

The fuel and hardware source regions of each assembly are homogenized within the volumes defined by the periphery of the fuel assembly and the source region axial extents. Cask body details include the true axial extent of the cask shield as described by the drawings in Section 1.3.2.

The geometric description of a MCBEND model is based on the Fractal Geometry combinatorial geometry system embedded in the code. In this system, bodies such as cylinders and rectangular parallelepipeds, and their logical intersections and unions, are used to describe the extent of material zones.

MCBEND employs an automated biasing technique for the Monte Carlo calculation based on a three-dimensional adjoint diffusion calculation. Mesh cells for the adjoint solution are selected based on half value thicknesses for each material.

MCBEND Monte Carlo calculations are performed for each source type present in each source region. This approach entails seven separate analyses, encompassing fuel neutron, fuel gamma, fuel n-gamma (secondary gammas arising from neutron interaction in the shield), fuel hardware, upper plenum, and upper and lower end-fitting gamma sources. Typically, a total of some 5 to 20 million histories are tracked to yield dose rate profiles for each model. These cases are analyzed for both radial and axial detector locations and for normal and hypothetical accident conditions. Similar analyses are also conducted for GTCC waste.

5.3.3.1 Connecticut Yankee Fuel Assembly Model

Based on the fuel parameters provided in Table 5.1-2, homogenized treatments of the stainless steel and Zircaloy clad fuel assembly source regions are developed. The homogenized fuel assembly is represented in the model as a stack of boxes with width equal to the fuel assembly width. The height of each box corresponds to the modeled height of the corresponding assembly region, shown in Figure 5.1-5.

The active fuel region homogenizations for stainless steel and Zircaloy clad fuel are shown in Tables 5.3-7 and 5.3-8, respectively. The interstitial material is void under the dry canister conditions of transport. The clad region is either stainless steel (density 7.92 g/cm³) or Zircaloy

(density 6.55 g/cm^3) depending on the fuel type. The resulting regional compositions on an atom/barn-cm basis are shown in Table 5.3-9.

Fuel assembly non-fuel regions are homogenized as shown in Tables 5.3-10 and 5.3-11 for stainless steel and Zircaloy clad fuel types, respectively. The only material included in the homogenized region is stainless steel. Volume fractions of material are based on the modeled regional volume and the volume of stainless steel present as computed from the modeled mass and density (7.92 g/cm^3). Although the geometry of the fuel end plug regions is modeled explicitly, the radiation source from activated fuel pin end plug material is assigned to adjacent source regions. For upper end plugs, the activation source term is assigned to the upper plenum source region, and for lower end plugs, the source term is assigned to the lower nozzle source region. This simplifying assumption reduces the number of source cases required and has little impact on computed dose rates.

5.3.3.2 CY-MPC Canister and Basket Model

For a given fuel type, the MCBEND description of the canister and basket elements forms a common sub-model employed in the transport cask analysis. The key features of the model are the detailed representation of fuel tubes, basket support and heat transfer disks, and weldment structures; the inclusion of the vent and drain ports in the canister shield lid; and explicit modeling of the shielding installed beneath the lid ports.

The vent and drain ports in the canister shield lid are modeled as a series of three stacked cylinders. The port cover is also modeled, and is in place for all transport analyses. The port cover is modeled as a solid piece of stainless steel. The CY-MPC canister is shown in Figure 5.3-10.

5.3.3.3 Description of CY GTCC Waste, Basket, and Canister Model

A homogenized treatment of CY GTCC waste is developed based on loading 800 lbs. of GTCC waste into each GTCC tube and calculating a volume fraction of stainless steel over the length of the GTCC fuel assembly size can (GTCC can) that holds the waste. The loading of 1300 lbs. in the center four tubes is accomplished by scaling the source strength in those locations internally in MCBEND. The calculated volume fraction based on a tube width of 8.75 in. and a GTCC can height of 133.50 in., is 0.2741.

The 24 stainless steel CY GTCC waste tubes are surrounded by a stainless steel cylindrical shell weldment that retains the GTCC waste and provides additional shielding. The CY GTCC basket support ribs are not modeled. The CY GTCC canister is modeled as identical to the fuel canister.

A schematic of the CY GTCC basket is shown in Figure 5.3-12.

5.3.3.4 Description of MCBEND NAC-STC Model

The three-dimensional model of the NAC-STC cask containing Connecticut Yankee design basis fuel assemblies or GTCC waste is based on the following features:

Normal conditions:

- Lower rotating trunnions
- Annular void below the neutron shield
- Radial neutron shield and shield shell
- Radial neutron shield heat fins
- All balsa upper and lower impact limiters (simplified to a length of 33.50 in.)

Accident conditions:

- Top axial lead slump
- Bottom axial lead slump
- Radial lead slump
- Loss of radial neutron shield and shield shell
- Loss of upper and lower impact limiters

Features common to both the normal and accident condition models are the inner lid vent and drain ports, the inner lid neutron shield, the bottom forging neutron shield, and the canister spacer below the CY-MPC canister.

Detailed model parameters used in creating the three-dimensional model are taken directly from the drawings in Section 1.3.2. Elevations associated with the transport cask three-dimensional features are established with respect to the center bottom of the canister for the MCBEND combinatorial model. The three-dimensional NAC-STC model is shown in Figure 5.3-11.

5.3.3.5 CY-MPC Shield Regional Densities

Based on the homogenization described in Section 5.3.3.1, the resulting active fuel regional densities are shown in Table 5.3-9 for stainless steel and Zircaloy clad fuel types. Material compositions for remaining structural and shield materials and GTCC waste are shown in Table 5.3-12. Compositions for fuel assembly non-fuel regions are equivalent to the stainless steel composition in Table 5.3-12 scaled by the material volume fractions shown in Tables 5.3-10 and 5.3-11.

Table 5.3-10

Assembly	Region	Material	Volume Fraction	Density (g/cc)	Weight Fraction
1A	1	Stainless Steel	0.95	7.8	0.74
1A	2	Zircaloy	0.05	6.5	0.03
1A	3	Concrete	0.01	2.3	0.01
1A	4	Water	0.01	1.0	0.01
1A	5	Steel	0.01	7.8	0.01
1A	6	Concrete	0.01	2.3	0.01

Figure 5.3-1 Three-Dimensional MCBEND Model for Directly Loaded Fuel – Normal
Conditions – Axial Detail



Figure Withheld Under 10 CFR 2.390

Figure 5.3-2 Three-Dimensional MCBEND Model for Directly Loaded Fuel – Accident
Conditions – Axial Detail

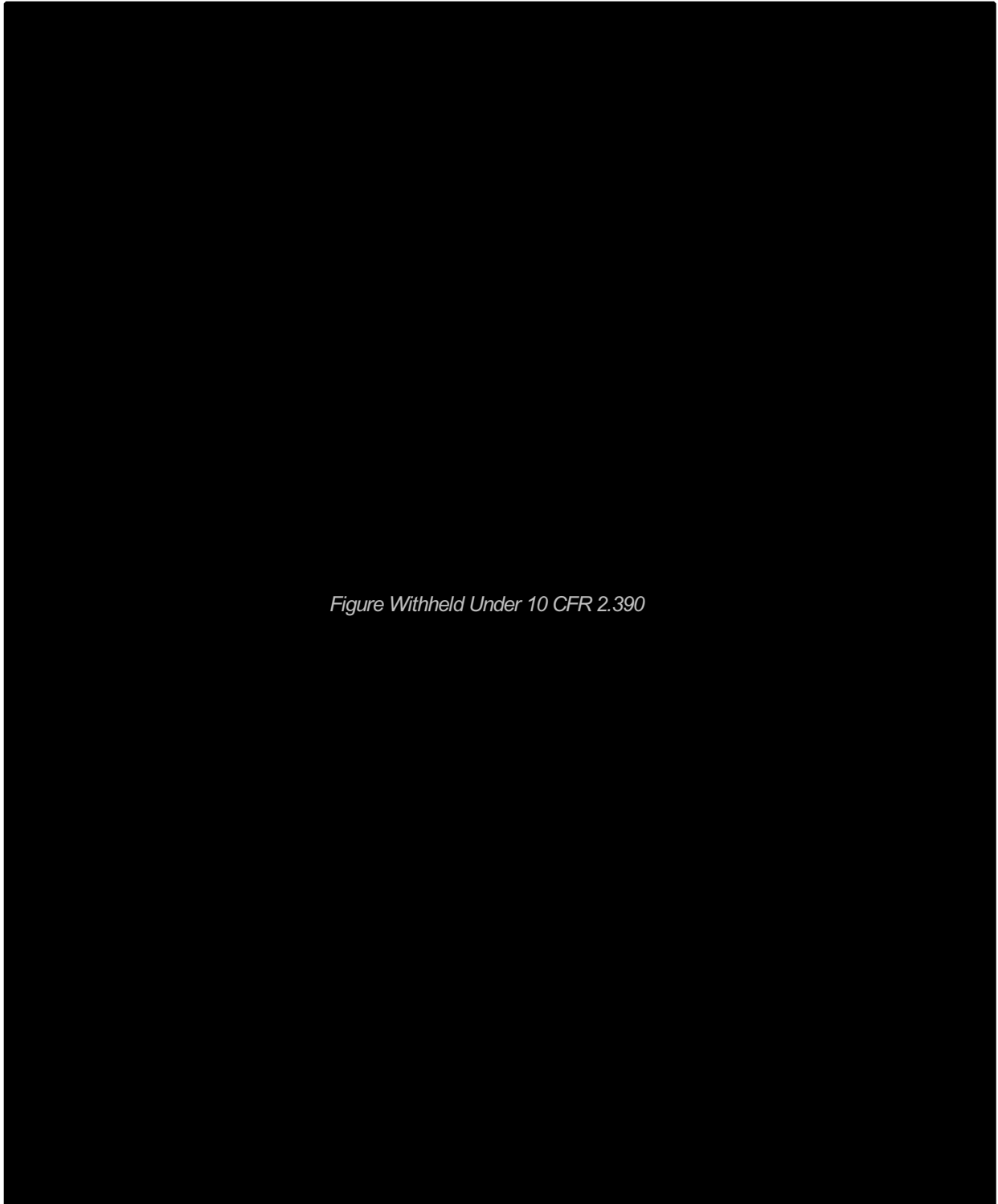


Figure Withheld Under 10 CFR 2.390

Figure 5.3-3 Three-Dimensional MCBEND Model for Directly Loaded Fuel – Radial Detail

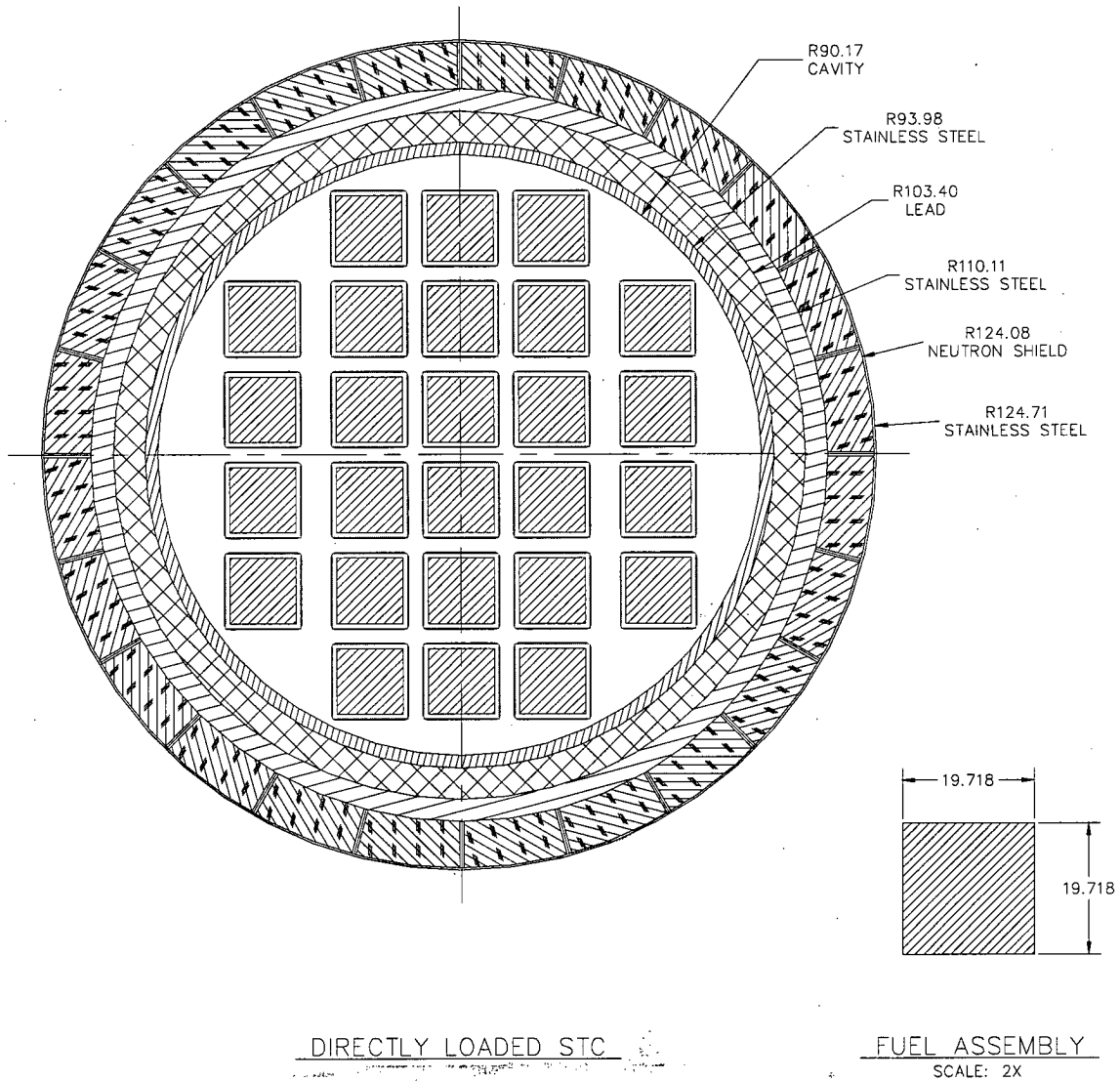


Figure 5.3-4 One-Dimensional Radial Shielding Model with Canistered Yankee Class Fuel

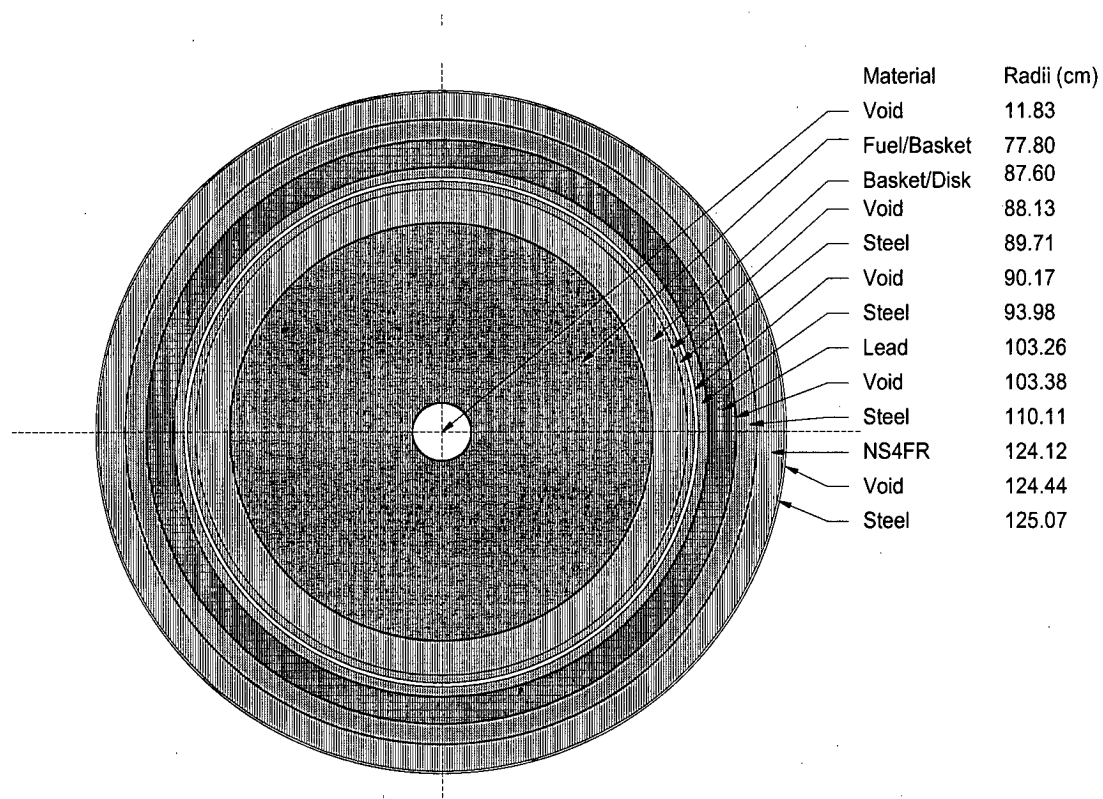


Figure 5.3-5 One-Dimensional Axial Shielding Model with Canistered Yankee Class Fuel

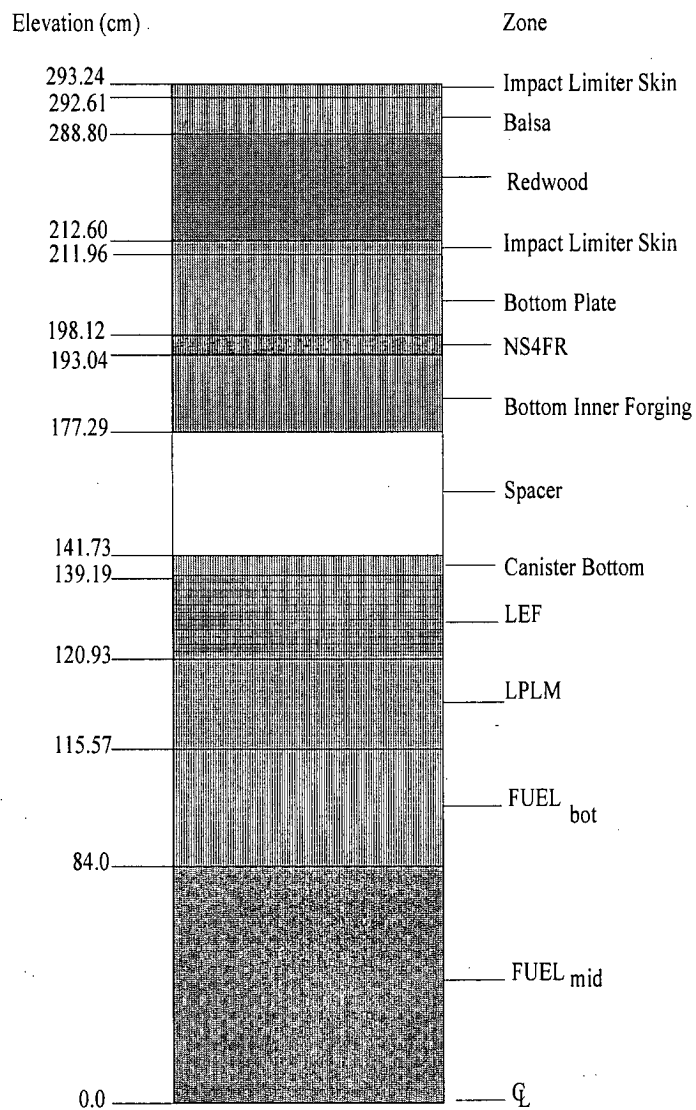


Figure 5.3-6 One-Dimensional Top Axial Model with Canistered Yankee Class Fuel

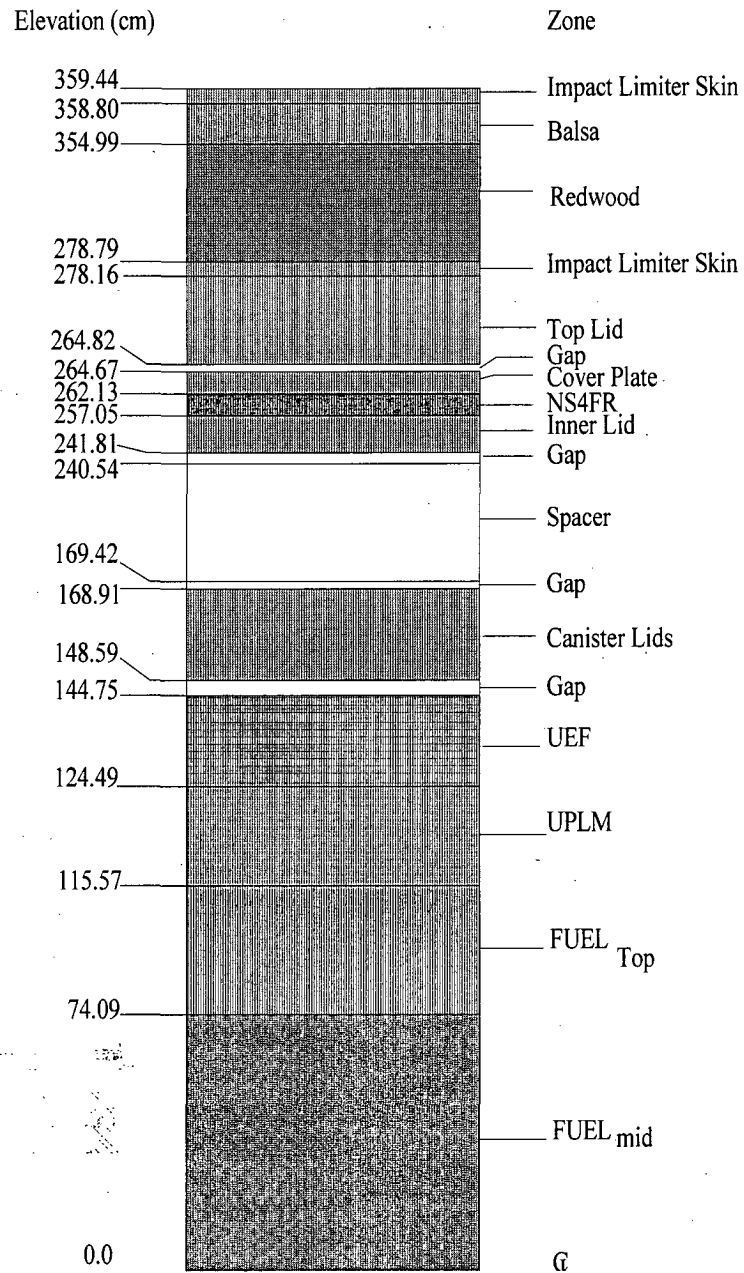


Figure 5.3-7 One-Dimensional Radial Shielding Model with Canistered Yankee GTCC Waste

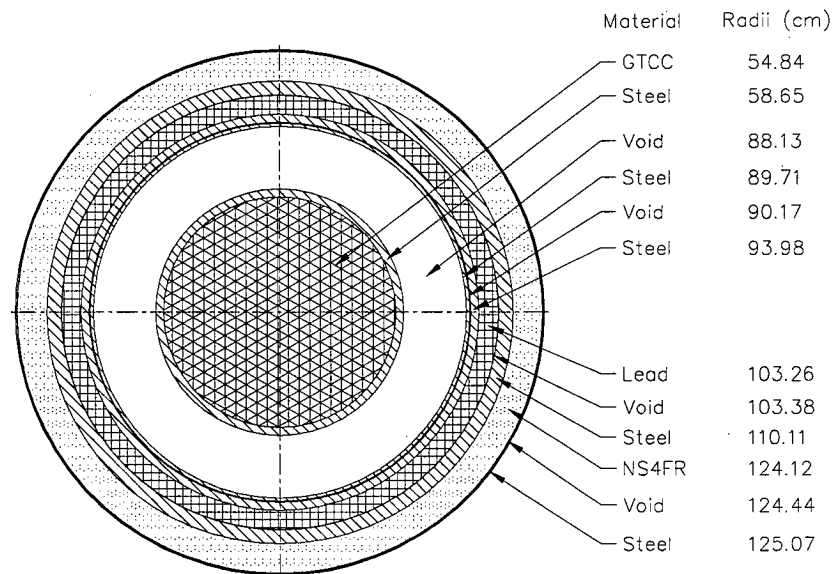


Figure 5.3-8 One-Dimensional Bottom Axial Model with Canistered Yankee GTCC Waste

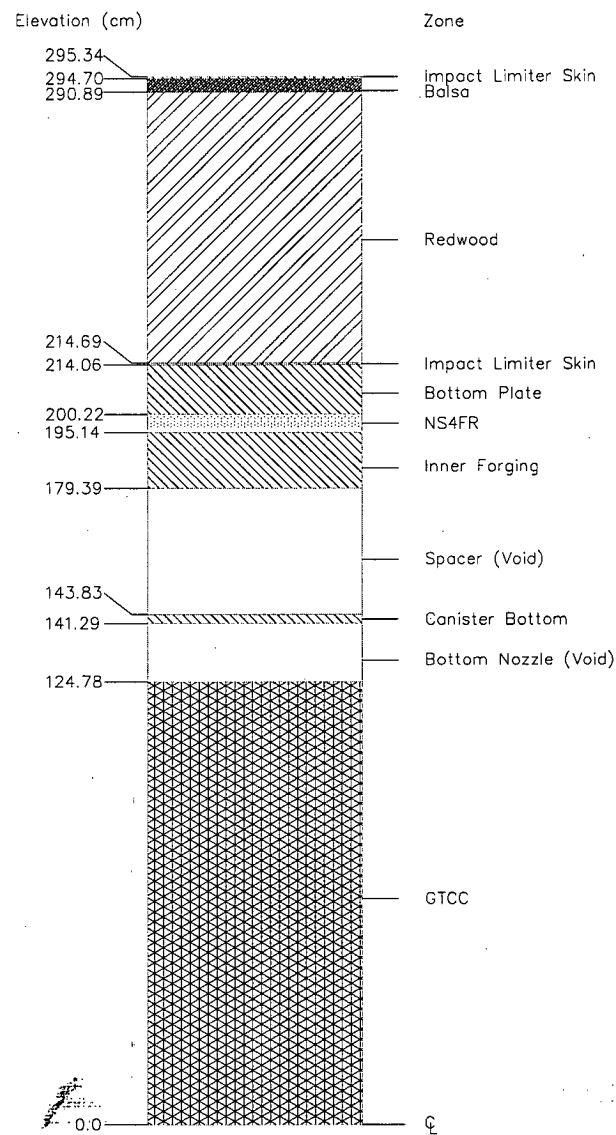


Figure 5.3-9 One-Dimensional Top Axial Model with Canistered Yankee GTCC Waste

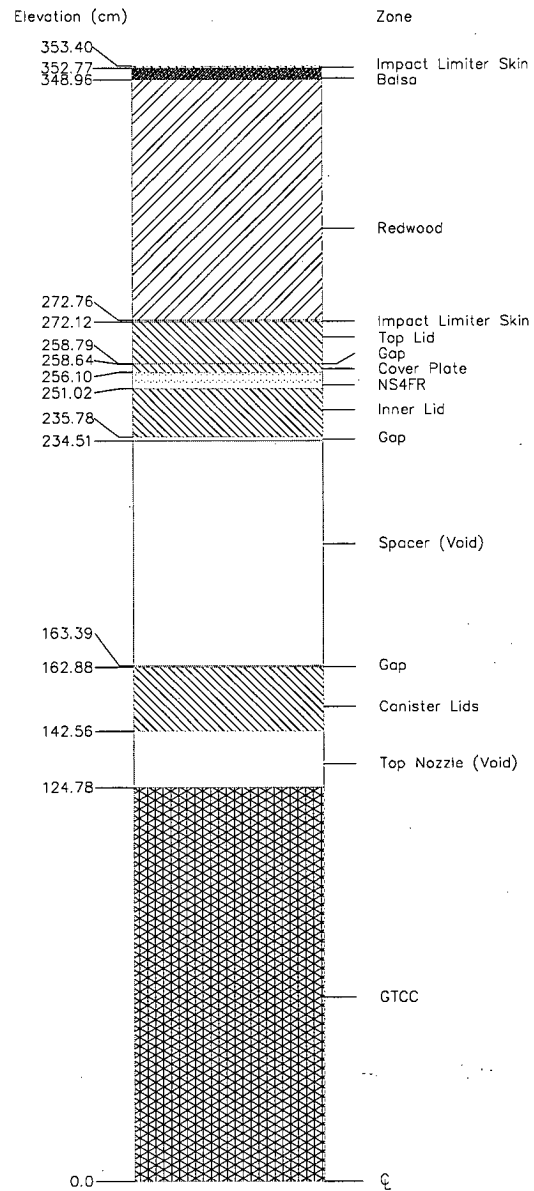


Figure 5.3-10 CY-MPC Three-Dimensional Canister Model Detail

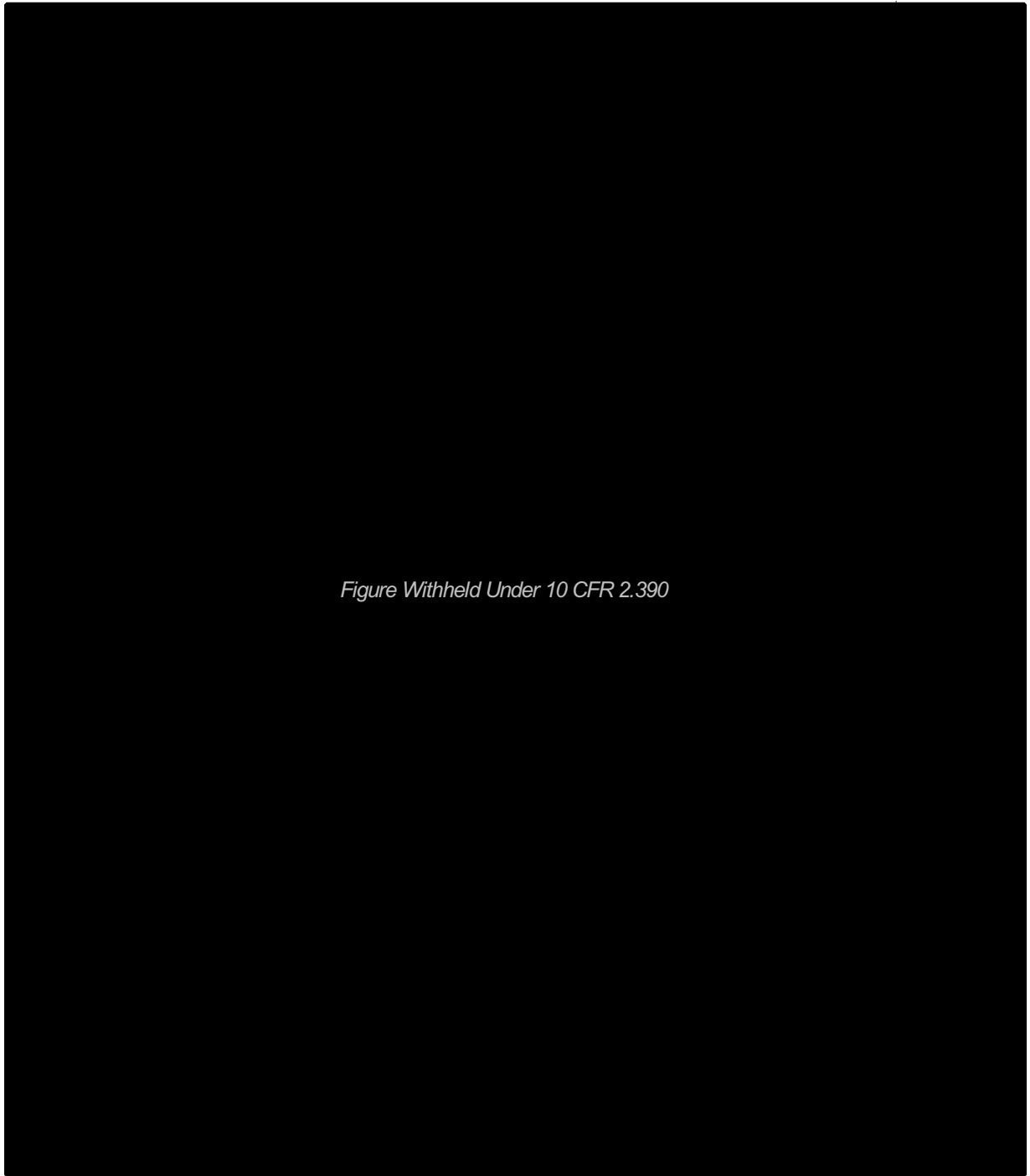


Figure Withheld Under 10 CFR 2.390

(Dimensions in cm)

Figure 5.3-11 Three-Dimensional NAC-STC Model for CY-MPC Analysis

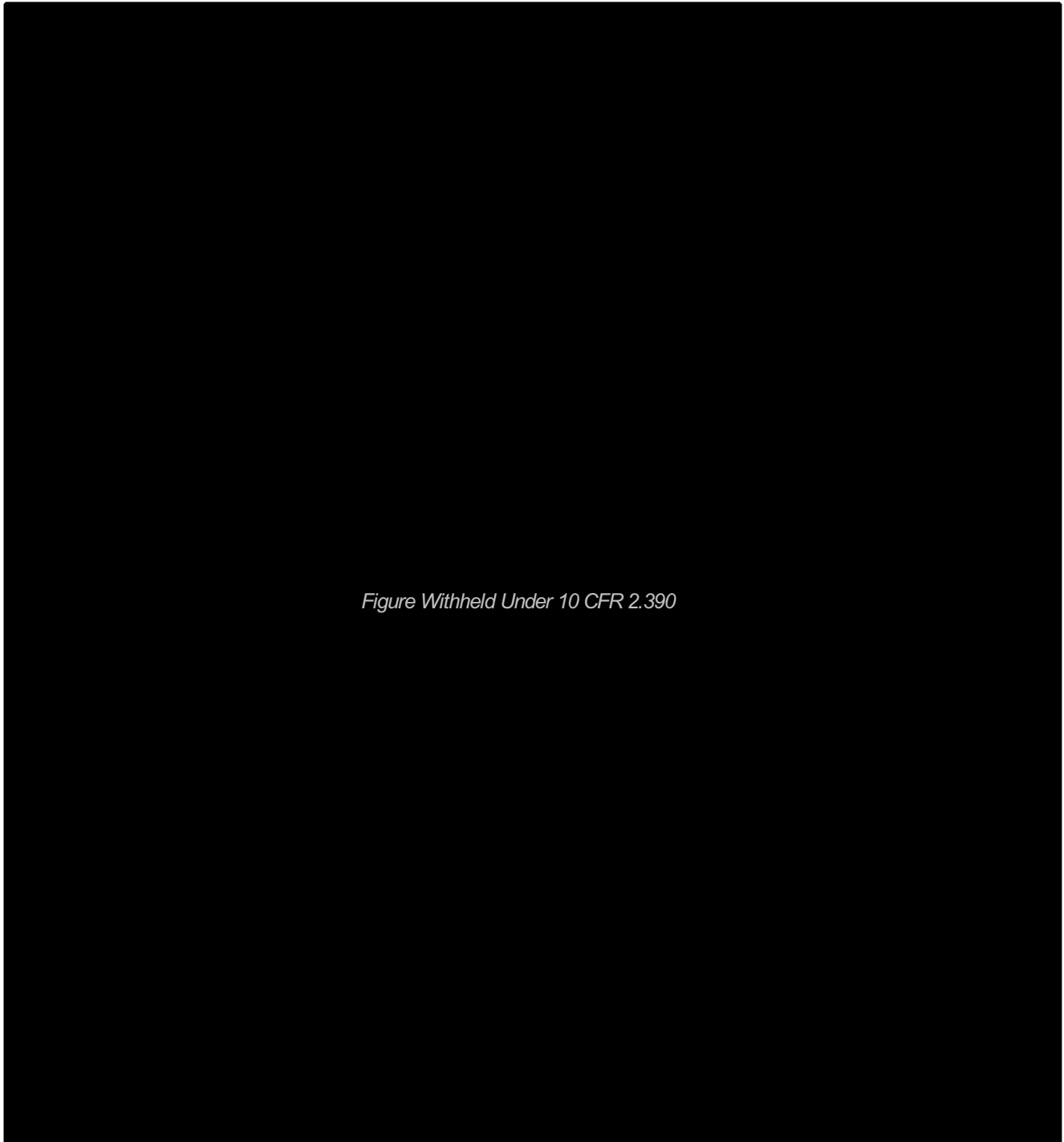


Figure Withheld Under 10 CFR 2.390

(Dimensions in cm)

Figure 5.3-12

Three-Dimensional Model of CY-MPC GTCC Waste Basket

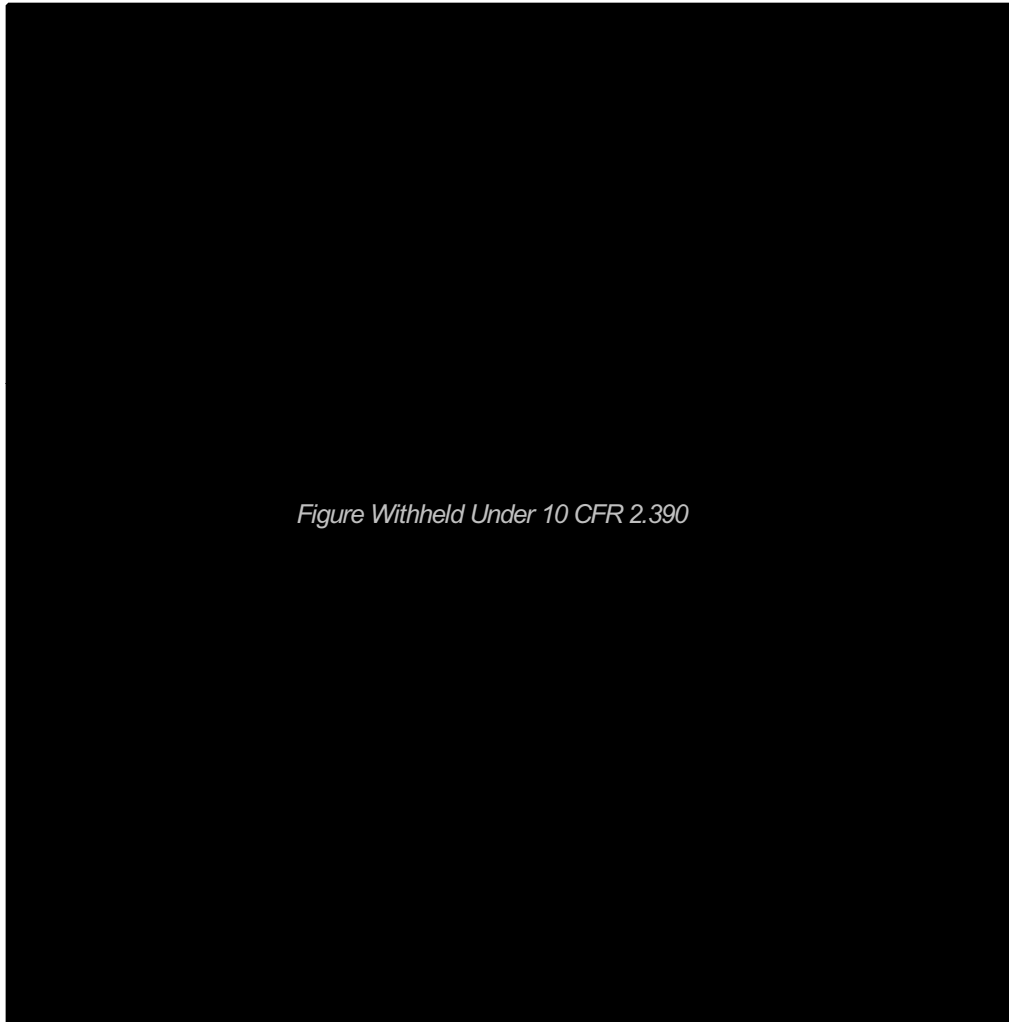


Figure Withheld Under 10 CFR 2.390

(Dimensions in cm)

Table 5.3-1 Directly Loaded Fuel Region Homogenization

Reference Fuel Type	Component	Volume Fraction of Components			
		UO ₂	Void	Clad	Interstitial
14×14	Fuel	3.1489E-01	--	--	--
	Gap	--	1.6671E-02	--	--
	Clad	--	--	8.3877E-02	--
	Guide Tube	--	--	1.2662E-02	--
	Instrument Tube	--	--	7.9139E-04	--
	Inside Tubes	--	--	--	3.7699E-02
	Interstitial	--	--	--	5.3341E-01
	Total	3.1489E-01	1.6671E-02	9.7331E-02	5.7111E-01
15×15	Fuel	3.0214E-01	--	--	--
	Gap	--	1.1135E-02	--	--
	Clad	--	--	8.6427E-02	--
	Guide Tube	--	--	6.1920E-03	--
	Instrument Tube	--	--	3.0960E-04	--
	Inside Tubes	--	--	--	4.7622E-02
	Interstitial	--	--	--	5.4618E-01
	Total	3.0214E-01	1.1135E-02	9.2929E-02	5.9380E-01
16×16	Fuel	2.9840E-01	--	--	--
	Gap	--	1.2993E-02	--	--
	Clad	--	--	1.0086E-01	--
	Guide Tube	--	--	5.4230E-03	--
	Instrument Tube	--	--	1.3558E-03	--
	Inside Tubes	--	--	--	6.7633E-02
	Interstitial	--	--	--	5.1334E-01
	Total	2.9840E-01	1.2993E-02	1.0763E-01	5.8097E-01
17×17	Fuel	3.0346E-01	--	--	--
	Gap	--	1.2613E-02	--	--
	Clad	--	--	9.2078E-02	--
	Guide Tube	--	--	7.3113E-03	--
	Instrument Tube	--	--	3.0464E-04	--
	Inside Tubes	--	--	--	5.4568E-02
	Interstitial	--	--	--	5.2967E-01
	Total	3.0346E-01	1.2613E-02	9.9694E-02	5.8424E-01

Table 5.3-2 Directly Loaded Fuel Homogenized Elemental Densities

Element	Density [atom/b-cm]			
	14x14	15x15	16x16	17x17
	(3.92 g/cm ³)	(3.75 g/cm ³)	(3.81 g/cm ³)	(3.79 g/cm ³)
Cr	7.38366E-06	7.04970E-06	8.16503E-06	7.56288E-06
Fe	1.37490E-05	1.31272E-05	1.52040E-05	1.40828E-05
Hf	2.15094E-07	2.05366E-07	2.37856E-07	2.20315E-07
Ni	6.54113E-07	6.24528E-07	7.23334E-07	6.69990E-07
O	1.46478E-02	1.40545E-02	1.38846E-02	1.40137E-02
Sn	4.85117E-05	4.63175E-05	5.36454E-05	4.96892E-05
U	7.31477E-03	7.01858E-03	6.93178E-03	6.99729E-03
Zr	4.12775E-03	3.94106E-03	4.56457E-03	4.22794E-03

Table 5.3-3 Directly Loaded Fuel Assembly Activated Hardware Region Homogenization

Reference Fuel Type	Region	Mass SS [kg/assy]	SS Volume [cm ³ /assy]	Height [cm]	Volume [cm ³ /assy]	Volume Fraction
14×14	Lower Nozzle	7.89	9.9659E+02	8.6944	3.3804E+03	2.9482E-01
	Upper Plenum	8.05	1.0164E+03	22.8016	8.8653E+03	1.1465E-01
	Upper Nozzle	9.89	1.2487E+03	8.8900	3.4564E+03	3.6128E-01
15×15	Lower Nozzle	5.68	7.1717E+02	10.7607	4.9559E+03	1.4471E-01
	Upper Plenum	4.12	5.2020E+02	21.2941	9.8070E+03	5.3044E-02
	Upper Nozzle	11.84	1.4949E+03	8.8392	4.0709E+03	3.6723E-01
16×16	Lower Nozzle	5.40	6.8182E+02	11.9685	5.0661E+03	1.3458E-01
	Upper Plenum	10.70	1.3510E+03	25.1358	1.0640E+04	1.2698E-01
	Upper Nozzle	9.50	1.1995E+03	17.3253	7.3336E+03	1.6356E-01
17×17	Lower Nozzle	6.31	7.9634E+02	10.4324	4.7831E+03	1.6649E-01
	Upper Plenum	5.41	6.8308E+02	25.6684	1.1768E+04	5.8043E-02
	Upper Nozzle	7.85	9.9116E+02	8.8392	4.0526E+03	2.4457E-01

Table 5.3-4 Directly Loaded Fuel Assembly Zircaloy Hardware Region Homogenization

Reference Fuel Type	Region	Mass Zirc [kg/assy]	Zirc Volume [cm ³ /assy]	Height [cm]	Volume [cm ³ /assy]	Volume Fraction
14×14	Lower Nozzle	1.84	2.8103E+02	8.6944	3.3804E+03	8.3137E-02
	Upper Plenum	4.98	7.6064E+02	22.8016	8.8653E+03	8.5800E-02
15×15	Lower Nozzle	2.10	3.2029E+02	10.7607	4.9559E+03	6.4628E-02
	Upper Plenum	6.99	1.0670E+03	21.2941	9.8070E+03	1.0880E-01
16×16	Lower Nozzle	2.59	3.9492E+02	11.9685	5.0661E+03	7.7953E-02
	Upper Plenum	7.97	1.2170E+03	25.1358	1.0640E+04	1.1439E-01
17×17	Lower Nozzle	2.79	4.2540E+02	10.4324	4.7831E+03	8.8938E-02
	Upper Plenum	6.85	1.0462E+03	25.6684	1.1768E+04	8.8896E-02

Table 5.3-5 Regional Densities for Directly Loaded Cask Structural and Shield Materials

Material	Element	Density [atom/b-cm]	Density [g/cm ³]
Stainless Steel	Cr	1.65112E-02	7.92
	Fe	6.31986E-02	
	Ni	6.50094E-03	
Zircaloy	Cr	7.58615E-05	6.55
	Fe	1.41261E-04	
	Hf	2.20993E-06	
	Ni	6.72052E-06	
	O	2.46540E-04	
	Sn	4.98421E-04	
	Zr	4.24095E-02	
Aluminum	Al	6.02626E-02	2.70
Lead	Pb	3.20871E-02	11.04
NS-4-FR	Al	7.80000E-03	1.63
	B	4.27500E-04	
	C	2.26000E-02	
	H	5.85000E-02	
	N	1.39000E-03	
	O	2.61000E-02	
Heat Fin	Cu	3.62309E-02	8.35
	Fe	3.61117E-02	
	Cr	9.43448E-03	
	Ni	3.71464E-03	
Balsa	C	2.78553E-03	0.125
	H	4.64261E-03	
	O	2.32135E-03	
NS-4-FR (Fire)	Al	7.80000E-03	0.81
	B	4.27500E-04	
	C	2.26000E-02	

Table 5.3-6 Yankee Class Fuel and Yankee GTCC Material Compositions

Zone/Material	Density (g/cc)	Nuclides	Density (atom/b-cm)
Middle Fuel Zone			
UO ₂	2.2769	²³⁴ U	2.79304E-07
		²³⁵ U	3.65635E-05
		²³⁸ U	5.04141E-03
		O	1.01565E-02
Zircaloy	0.6417	Zr	4.23638E-03
SS304	0.3420	Cr	7.52598E-04
		Mn	7.49779E-05
		Fe	2.56318E-03
		Ni	3.33394E-04
Aluminum	0.1091	Al	2.43503E-03
B ₄ C	0.0101	¹⁰ B	8.76394E-05
		¹¹ B	3.52760E-04
		C	1.10100E-04
Middle Basket/Disk Zone			
SS304	0.9543	Cr	2.10001E-03
		Mn	2.09215E-04
		Fe	7.15218E-03
		Ni	9.30286E-04
Aluminum	0.2920	Al	6.51722E-03
Top Fuel/Basket Zone			
UO ₂	2.2769	²³⁴ U	2.79304E-07
		²³⁵ U	3.65635E-05
		²³⁸ U	5.04141E-03
		O	1.01565E-02
Zircaloy		Zr	4.04558E-03
SS304	0.3084	Cr	6.78658E-04
		Mn	6.76116E-05
		Fe	2.31136E-03
		Ni	3.00639E-04
Aluminum	0.0622	Al	1.38826E-03
B ₄ C	0.0101	¹⁰ B	8.76394E-05
		¹¹ B	3.52760E-04
		C	1.10100E-04
Top Plenum Zone			
Zircaloy	0.5718	Zr	3.77491E-03
SS304	0.4821	Cr	1.48517E-03
		Mn	1.05692E-04
		Fe	3.61319E-03
		Ni	4.69968E-04

Table 5.3-6 Yankee Class Fuel and Yankee GTCC Material Compositions (Continued)

Material/Zone	Density (g/cc)	Nuclides	Density (atom/b-cm)
Top End Fitting Zone			
SS304	0.6749	Cr	1.74286E-02
		Mn	1.47961E-04
		Fe	5.05816E-03
		Ni	6.57917E-04
Bottom Fuel/Basket Zone			
UO ₂	2.2769	²³⁴ U	2.79304E-07
		²³⁵ U	3.65635E-05
		²³⁸ U	5.04141E-03
		O	1.01565E-02
Zircaloy	0.6128	Zr	4.04558E-03
SS304	0.2350	Cr	6.49170E-04
		Mn	6.46739E-05
		Fe	2.21093E-03
		Ni	2.87577E-04
Aluminum	0.0622	Al	1.38826E-03
B ₄ C	0.0101	¹⁰ B	8.76394E-05
		¹¹ B	3.52760E-04
		C	1.10100E-04
Bottom Plenum Zone			
Zircaloy	0.6128	Zr	4.0455E-03
SS304	1.0529	Cr	2.31699E-03
		Mn	2.30831E-03
		Fe	7.89115E-03
		Ni	1.02640E-03
Bottom Endfitting Zone			
SS304	0.9664	Cr	2.12664E-03
		Mn	2.11867E-04
		Fe	7.24287E-03
		Ni	9.42082E-04
GTCC Waste and Container			
SS304	3.29	Cr	7.2399E-03
		Mn	7.2128E-04
		Fe	2.4658E-02
		Ni	3.2072E-03

Table 5.3-7 Connecticut Yankee Stainless Steel Clad Fuel Region Homogenization

Component	Volume Fraction of Components			
	UO ₂	Void	Clad	Interstitial
Fuel	3.3010E-01	--	--	--
Gap	--	9.5568E-03	--	--
Clad	--	--	6.0055E-02	--
Guide Tube	--	--	5.6120E-03	--
Instrument Tube	--	--	2.9439E-04	--
Inside Tubes	--	--	--	6.0931E-02
Interstitial	--	--	--	5.3345E-01
Total	3.3010E-01	9.5568E-03	6.5962E-02	5.9438E-01

Table 5.3-8 Connecticut Yankee Zircaloy Clad Fuel Region Homogenization

Component	Volume Fraction of Components			
	UO ₂	Void	Clad	Interstitial
Fuel	3.0394E-01	--	--	--
Gap	--	9.0410E-03	--	--
Clad	--	--	8.9574E-02	--
Guide Tube	--	--	7.7242E-03	--
Instrument Tube	--	--	3.8621E-04	--
Inside Tubes	--	--	--	5.7525E-02
Interstitial	--	--	--	5.3181E-01
Total	3.0394E-01	9.0410E-03	9.7684E-02	5.8934E-01

Table 5.3-9 Connecticut Yankee Homogenized Fuel Regional Densities

Stainless Steel Clad		Zircaloy Clad	
Element	Density [atom/b-cm]	Element	Density [atom/b-cm]
Cr	1.08911E-03	Cr	7.41045E-06
Fe	4.16871E-03	Fe	1.37989E-05
Ni	4.28815E-04	Hf	2.15875E-07
O	1.53301E-02	Ni	6.56487E-07
U	7.66861E-03	O	1.41392E-02
--	--	Sn	4.86877E-05
--	--	U	7.06088E-03
--	--	Zr	4.14273E-03

Table 5.3-10 Connecticut Yankee Stainless Steel Clad Fuel Assembly Hardware Region Homogenization

Region	Mass SS [kg/cask]	SS Volume [cm ³]	Height [cm]	Volume [cm ³]	Volume Fraction
Lower Nozzle	230.10	2.9053E+04	8.0975	9.6963E+04	2.9963E-01
Lower End Plug	65.96	8.3277E+03	1.7399	2.0834E+04	3.9971E-01
Fuel Hardware	2673.64	3.3758E+05	309.3720	3.7045E+06	9.1126E-02
Upper Plenum	100.85	1.2734E+04	9.0162	1.0796E+05	1.1795E-01
Upper End Plug	65.96	8.3277E+03	1.7399	2.0834E+04	3.9971E-01
Upper Nozzle	292.24	3.6899E+04	17.2466	2.0652E+05	1.7867E-01

Table 5.3-11 Connecticut Yankee Zircaloy Clad Fuel Assembly Hardware Region
Homogenization

Region	Mass SS [kg/cask]	SS Volume [cm ³]	Height [cm]	Volume [cm ³]	Volume Fraction
Lower Nozzle	141.44	1.7859E+04	8.0975	9.6963E+04	1.8418E-01
Lower End Plug	0.0	0.0000E+00	1.7399	2.0834E+04	0.0000E+00
Fuel Hardware	504.79	6.3736E+04	307.5940	3.6832E+06	1.7304E-02
Upper Plenum	133.56	1.6864E+04	10.7942	1.2925E+05	1.3047E-01
Upper End Plug	0.0	0.0000E+00	1.7399	2.0834E+04	0.0000E+00
Upper Nozzle	307.84	3.8869E+04	17.2466	2.0652E+05	1.8821E-01

Table 5.3-12 Regional Densities for CY-MPC Structural and Shield Materials

Material	Element	Density [atom/b-cm]
Stainless Steel	Cr	1.65112E-02
	Fe	6.31986E-02
	Ni	6.50094E-03
BORAL Aluminum Clad	Al	3.35910E-02
	B	4.63378E-02
	C	1.21776E-02
Aluminum	Al	6.02626E-02
Lead	Pb	3.20871E-02
NS-4-FR	Al	7.80000E-03
	B	4.27500E-04
	C	2.26000E-02
	H	5.85000E-02
	N	1.39000E-03
	O	2.61000E-02
Heat Fin	Cu	3.62309E-02
	Fe	3.61117E-02
	Cr	9.43448E-03
	Ni	3.71464E-03
Balsa	C	2.78553E-03
	H	4.64261E-03
	O	2.32135E-03
GTCC Waste	Cr	4.52571E-03
	Fe	1.73227E-02
	Ni	1.78191E-03

5.4 Shielding Evaluation

The techniques used to perform gamma and neutron dose rate calculations for the NAC-STC in the directly loaded and canistered configurations are described below, including descriptions of the computer codes and methods that were used in the shielding analyses of the cask.

5.4.1 Computer Code Descriptions and Results

5.4.1.1 Directly Loaded Fuel Configuration

To calculate three-dimensional NAC-STC dose rates and define a minimum cool time table, a response function approach is taken to the shielding evaluation. Instead of running numerous direct-solution cases at a discrete matrix of burnups, enrichments, and cool times, three-dimensional dose rate response function cases are executed, thereby evaluating the dose contribution of a "unit" source in each significant energy group in each source region. Six sets of response functions are required based on the defined source regions: fuel gamma, fuel neutron, fuel n-gamma, lower end-fitting gamma, upper plenum gamma and upper end-fitting gamma. Fuel hardware contributions to dose rates are evaluated using the fuel gamma response functions. For each source region, a subset of the 28 neutron and 22 gamma energy groups is evaluated to increase the efficiency of the analysis. Fuel gamma doses are calculated using gamma groups 7 through 16; fuel neutron and fuel n-gamma doses are calculated using neutron groups 2 through 15; and hardware sources are calculated using gamma groups 12 through 14. Neutron groups 1 and 16 through 28 contain no neutron source. The hardware source is dominated by the ^{60}Co source, which resides in groups 12 through 14. Fuel gamma groups 7 through 16 are selected based on the integrated source energy in each group (Source magnitude drops by a factor of 1,000 when moving from fuel group 7 to group 6.). The choice of energy groups evaluated is verified by running "direct" solutions using the complete spectrum and demonstrating that the total dose is reproduced within the Monte Carlo error of the analysis. Sample comparisons of the dose response method (DRM) to direct-solution cases are shown in Figures 5.4-7 and 5.4-8 for normal and accident conditions, respectively. As shown, the dose response method reproduces direct solution results within the uncertainty of the analysis. Since the dose response method evaluates each source energy line individually, and therefore avoids statistical sampling within the energy group structure of the source, the results of the dose response method tend towards smaller Monte Carlo uncertainty bands.

Calculational Methods

The shielding evaluation of the directly loaded configuration is performed using MCBEND. Source terms include fuel neutron, fuel gamma and gamma contributions from activated hardware. As described in Section 5.2.1.3, the evaluation includes the effect of fuel burnup peaking on fuel neutron and gamma source terms.

The MCBEND shielding model described in Section 5.3.1 is utilized with the source terms described in Section 5.2.1 to estimate the dose rate profiles at various distances from the side, top and bottom of the cask for both normal and accident conditions. The method of solution is continuous energy Monte Carlo with an adjoint diffusion solution for generating importance meshes. Radial biasing is performed within the MCBEND code to estimate dose rates on the side of the cask. Axial biasing is performed to estimate dose rates on the top and bottom of the cask.

The MCBEND code has been validated against various classical shielding problems, including fast and thermal neutron sources penetrating through single material slab geometries of iron, graphite and water. The validation suite also includes fast neutron transmission through alternating slabs of iron and water. Of particular interest is a benchmark of MCBEND to gamma and neutron dose rates outside a metal transport cask, where agreement between measurement and calculation is within 20% for the majority of dose locations.

MCBEND results are calculated using the JEF2.2 neutron cross-section library and the ANSWERS gamma library.

MCBEND Flux-to-Dose Conversion Factors

The ANSI/ANS 6.1.1-1977 flux-to-dose rate conversion factors are used in all shielding evaluations for directly loaded fuel. The ANSI/ANS gamma and neutron dose conversion factors are shown in Table 5.4-1 and Table 5.4-2. The number of energy/conversion factor pairs was increased to 133 neutron and 371 gamma pairs by a log-log interpolation scheme indicated as appropriate in ANSI/ANS 6.1.1-1977.

Loading Table for Directly Loaded Fuel

Three-dimensional radial response functions are generated for PWR fuel assemblies for both normal and accident conditions. Based on preliminary analysis, two bounding axial shift scenarios have been established: 1) maximum fuel assembly shift upward in the cask cavity without a corresponding shift in the basket and 2) no fuel assembly or basket shift. For axial

biasing, the limiting shift scenario corresponds exactly to the position of the fuel assembly in the cavity, i.e., top dose rates are maximized when the fuel assembly is shifted up and bottom dose rates are maximized when the fuel assembly is as far down in the cavity as possible. For radial biasing, the two different shift scenarios are limiting for different transport conditions. The maximum fuel assembly axial shift is limiting for normal conditions because upper plenum and upper end fitting hardware move adjacent to the location in the radial shield where the radial lead shield ends. The limiting shift is downward for accident conditions due to the bottom axial lead slump, which is adjacent to the lower end fitting hardware source.

The first step in determining limiting PWR dose rates for the directly loaded cask is the generation of dose rate response functions for generation of minimum cool time tables. For each array size, at each of 4 burnup, 15 enrichment, and 18 cool time combinations, dose rate profiles are calculated for both normal and accident transport conditions. Using these dose rate profiles, the maximum radial dose rates at 2 meters from the railcar are tabulated for normal conditions.

Minimum cool times are calculated to ensure that a decay heat limit of 850 W/assembly is not exceeded and that the dose rate at 2 meters from the railcar does not exceed 9.5 mrem/hr. The 9.5 mrem/hr analysis limit was chosen to provide margin against the 10 mrem/hr regulatory limit. Cool times needed to reach these limits are calculated using linear interpolation on the entire array of maximum dose rates. The linear interpolation is valid because of the exponential decrease in source term and, thus, dose rate as a function of time. The interpolated cool time is rounded up to the next integer year. A sample minimum cool time generation for the 14×14 reference assembly at 40,000 MWD/MTU is shown in Table 5.4-3. Repeating this analysis for all fuel types and burnups results in the complete loading table shown in Table 5.4-5. Based on the loading table, maximum radial dose rates for each fuel type are shown in Table 5.4-4.

The minimum cool times are used to calculate maximum accident condition dose rates at 1 meter from the cask. The 1000 mrem/hr limit is not exceeded at any of the calculated minimum cool times.

Based on the radial dose rate results for normal and accident conditions and their application to the minimum cool time table, the 14×14 reference assembly provides maximum dose rates. Thus, top axial and bottom axial response functions have been executed for this assembly only. This ensures that the maximum axial dose rates for the directly loaded system are captured, and that variations in burnup, enrichment and minimum cool time are thoroughly examined.

A summary of the limiting source terms for each transport condition and detector biasing is given below. All limiting source terms are taken from the 14×14 reference fuel assembly.

Detector Biasing	Normal Conditions			Accident Conditions		
	Burnup [MWD/MTU]	Enrichment [wt % ²³⁵ U]	Cool Time [Years]	Burnup [MWD/MTU]	Enrichment [wt % ²³⁵ U]	Cool Time [Years]
Radial	40,000	2.3	10	45,000	2.3	14
Top Axial	30,000	2.3	6	45,000	2.3	14
Bottom Axial	40,000	2.3	10	45,000	2.3	14

Three-Dimensional Dose Rates for Directly Loaded Fuel

Further detail on the three-dimensional dose rates are presented in Figures 5.4-1 through 5.4-6 for the limiting 14×14 reference assembly. Maximum dose rates are tabulated in Tables 5.4-6 and 5.4-7.

The maximum normal conditions surface dose rate is 366 mrem/hr at an axial elevation between the radial neutron shield and the upper impact limiter. At 1 meter from the surface of the neutron shield shell, the maximum dose rate is 20.3 mrem/hr. This dose rate defines the transport index. The maximum normal conditions dose rate at 2 meters from the cask railcar is 9.5 mrem/hr and occurs at an axial elevation adjacent to the upper plenum and upper end-fitting elevations. The maximum accident conditions dose rate at 1 meter from the cask is 665 mrem/hr and occurs at the cask midplane. The top and bottom axial dose rates are small when compared to the radial dose rate for the same transport conditions.

Dose rate variations from heat fins in the neutron shield are examined explicitly using azimuthal detectors that span the entire length of the neutron shield. As shown in Figure 5.4-4, peaks in the neutron dose rate correspond to dips in the gamma dose rate, and vice versa. Thus, the neutron dose rate increase resulting from the ducting is offset by the reduction of the gamma dose rate resulting from the additional shielding provided by the fins.

Detector descriptions for dose rates on the side of the STC are given in Tables 5.4-8 and 5.4-9 for normal and accident conditions, respectively. Note that an axial height of 0.0 cm corresponds to the bottom of the STC cavity.

5.4.1.2 Canistered Yankee Class Fuel and GTCC Waste

Shielding evaluations of canistered Yankee Class fuel and GTCC waste are performed with SCALE 4.3 for the PC (ORNL, 1995). In particular, SCALE 4.3 shielding analysis sequence SAS2H (Herman, 1995) is used to generate source terms for the design basis fuel and GTCC waste hardware and SAS1 (Knight, 1995) is used to perform one-dimensional radial and axial shielding analysis. Transverse leakage is accounted for by the use of radial and axial bucklings. The 27 group neutron, 18 group gamma, coupled cross section library (27N-18COUPLE) based on ENDF/B-IV (Jordan, 1995) is used in all shielding evaluations. Fuel source terms include fuel neutron, fuel gamma, and activated hardware gamma. GTCC waste hardware source terms are based on core baffle activated hardware characterization from dose rate measurements and baffle material chemical assay. Dose rate evaluations include the effect of fuel burnup peaking on fuel neutron and gamma source terms.

For the four Yankee Class fuel types, a nominal decrease in enrichment is evaluated to account for tolerances applied to enrichment specifications during fabrication. The evaluation shows that a small reduction in the batch-averaged minimum enrichments does not significantly affect calculated dose rates as shown by the minimal increase in source strength:

Fuel Type	Minimum Enrichment (wt % ²³⁵ U)	Percent Increase in Source (%)			
		Decay Heat	Neutron	Gamma	Hardware
CE	3.66	0.2	2.2	0.1	0.6
Exxon	3.46	0.2	2.2	0.0	0.7
UN	3.96	0.1	2.2	0.1	0.6
WE	4.90	0.1	2.2	0.0	0.5

5.4.1.3 Canistered CY-MPC Fuel and GTCC Waste

The shielding evaluations of the NAC-STC with canistered CY-MPC fuel or GTCC waste are performed with MCBEND version 9E (MCBEND). For the fuel evaluations, source terms include fuel neutron, fuel gamma and gamma contributions from activated hardware. As described in Section 5.2.3.4, these evaluations include the effect of fuel burnup peaking on fuel neutron and gamma source terms. The resulting dose rate profiles are reported as a function of distance from the radial and axial surfaces of the NAC-STC cask.

The MCBEND shielding models described in Section 5.3.3 are utilized with the source terms described in Section 5.2.3 to estimate the dose rate profiles along the surfaces of the transport cask. The method of solution is continuous energy Monte Carlo with an adjoint diffusion solution for generating importance meshes. Radial biasing is performed within the MCBEND code to estimate dose rates on the side of the transport cask, and axial biasing is performed to estimate dose rates on the top and bottom surfaces of the cask.

MCBEND Flux-to-Dose Conversion Factors

The ANSI/ANS 6.1.1-1977 flux-to-dose rate conversion factors are used in all CY-MPC shielding evaluations. Tables 5.4-10 and 5.4-11 show the regrouped flux-to-dose conversion factors on the MCBEND standard 28 group neutron and 22 group gamma energy boundaries.

CY-MPC Three-Dimensional Dose Rates

The CY-MPC three-dimensional model dose rates are presented in Figures 5.4-9 through 5.4-14 for the stainless steel clad fuel at 38,000 MWD/MTU and 10-year cool time and the Zircaloy clad fuel at 43,000 MWD/MTU and 10-year cool time. CY GTCC three-dimensional model dose rates are presented in Figures 5.4-15 and 5.4-16 for the design basis core baffle source at 10-year cool time. Dose rates at specified detector locations are presented in Tables 5.4-12 through 5.4-19 for stainless steel and Zircaloy clad fuel. For GTCC waste, the maximum and average dose rates at specified detector locations are presented in Tables 5.4-20 through 5.4-23.

For the design basis fuel sources, the maximum normal conditions surface dose rate is 49 mrem/hr at an axial elevation between the radial neutron shield and the upper impact limiter. The maximum normal conditions dose rate at two meters from the cask railcar is 3.6 mrem/hr and occurs at the cask midplane. The maximum accident conditions dose rate at one meter from the cask is 369 mrem/hr and occurs at the cask midplane. The top and bottom axial dose rates are small when compared to the radial dose rate for the same transport conditions.

Damaged fuel dose rates are calculated using the design basis Zircaloy clad fuel as a basis and filling the void regions in the fuel assembly non-fuel regions with UO_2 . This moves a significant neutron and gamma source close to the positions of least shielding. For damaged fuel sources, the maximum normal conditions surface dose rate increases to 110 mrem/hr, also at the gap between the radial neutron shield and the upper impact limiter. The maximum normal conditions dose rate at two meters from the cask increases to 3.8 mrem/hr. The maximum accident conditions dose rate at one meter from the cask increases to 376 mrem/hr. An increase in the top

axial and bottom axial dose rates is observed; however, the top and bottom axial dose rates remain small when compared to the radial dose rates.

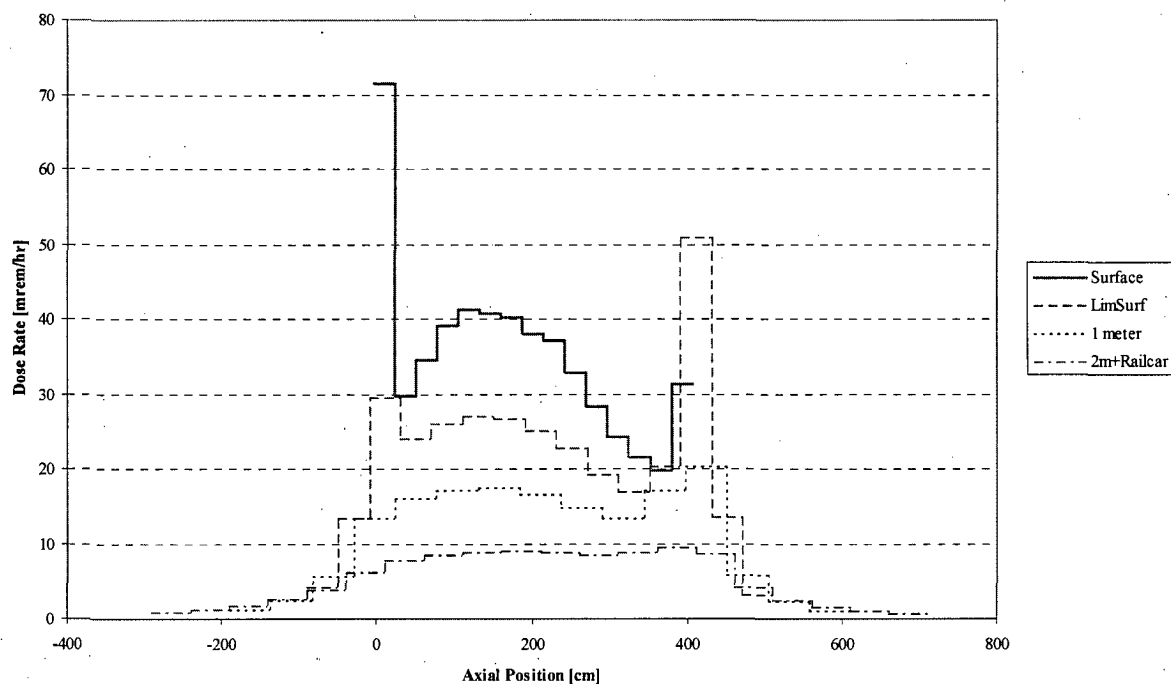
For the design basis GTCC waste, the maximum normal conditions radial surface dose rate is 6.1 mrem/hr and occurs at the cask midplane (the bottom axial surface dose rate is 4.2 mrem/hr). The maximum normal conditions dose rate at two meters from the cask is 1.5 mrem/hr and also occurs at the cask midplane (the bottom axial dose rate is 0.7 mrem/hr). The maximum accident conditions dose rate at one meter from the cask is 25.0 mrem/hr and occurs at the axial elevation of the modeled bottom axial lead slump due to gamma streaming and the minimal shielding of the canister bottom plate. Top axial dose rates are small in comparison to the radial and bottom axial dose rates because of the stainless steel lid structure.

The dose rate contributions from the activated Reactor Control Cluster Assemblies are small compared to those from the fuel sources. A full complement of 26 Reactor Control Cluster Assemblies cooled to a minimum of 10 years leads to an additional maximum radial surface dose rate of 0.6 mrem/hr for normal conditions and 3.2 mrem/hr for accident conditions at an axial position adjacent to the bottom of the fuel. The contribution to the Reactor Control Cluster Assemblies to the dose rate at an axial position adjacent to the fuel midplane is less than 1 mrem/hr for both normal and accident conditions. Hence, no significant increase in cask radial maximum dose rate occurs due to the inclusion of the cluster assemblies. Since the activated portion of the Reactor Control Cluster Assemblies resides in the bottom of the fuel, no significant increase in top axial dose rates is observed.

Additional dose rates due to inclusion of up to eight activated flow mixers inserted into the top nozzles of fuel assemblies in the center-most basket locations (positions 7, 8, 12, 13, 14, 15, 19, and 20) is small. The radial surface dose rate contribution from the flow mixers is a maximum of 0.4 mrem/hr for normal conditions and 7.2 mrem/hr for accident conditions. The top axial surface dose rate contribution from the flow mixers is less than 1 mrem/hr for both normal and accident conditions. The small dose rate increase is due to the effective self-shielding of the material by the surrounding fuel assemblies and the limited radial extent of the material with respect to the canister lids.

The dose rate increase due to the inclusion of up to two assemblies with a maximum of two irradiated stainless steel filler rods per assembly is less than 1% of the total dose rate at any surface of the NAC-STC. Assemblies containing irradiated stainless steel rods may only be loaded in basket positions 13 and 14, as shown in Figure 6.3-3.

Figure 5.4-1 Radial Dose Rate Profiles for Directly Loaded Fuel in Normal Conditions of Transport



Note: The dose rate at the surface of the cask between the neutron shield and the upper impact limiter is 366.3 (0.2%) mrem/hr.

Figure 5.4-2 Radial Dose Rate Profile by Source Type at 2 meters from the Railcar for Directly Loaded Fuel in Normal Conditions of Transport

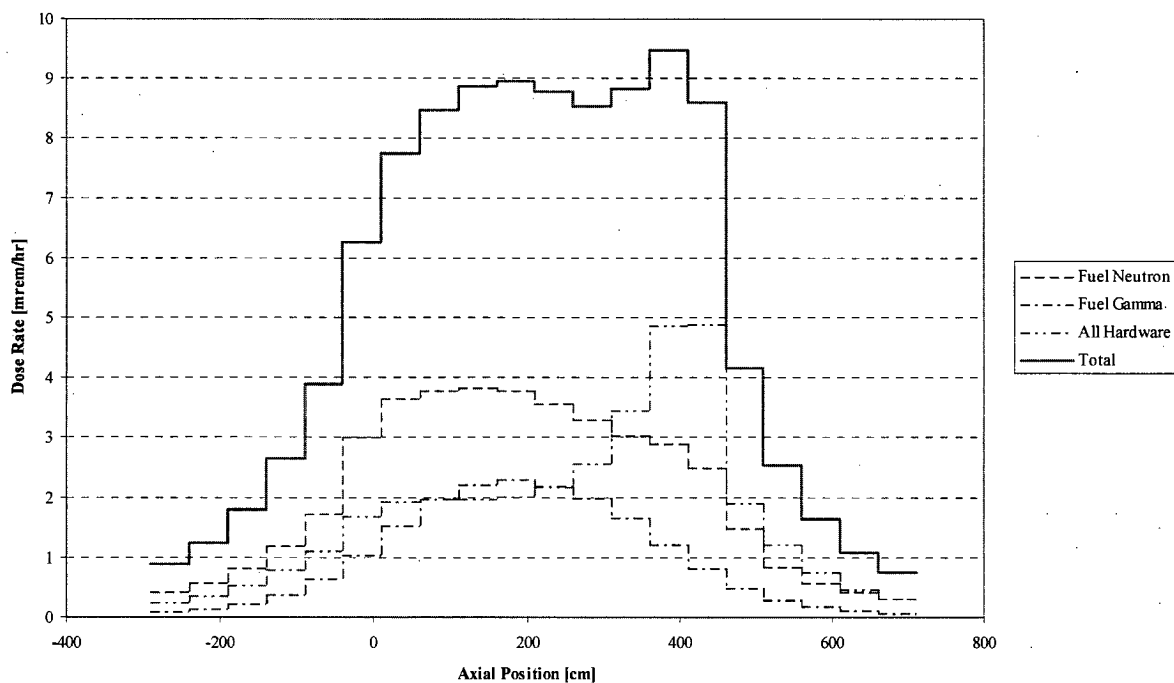


Figure 5.4-3 Azimuthal Radial Surface Dose Rate Profile by Source Type at Rotation Trunnion Elevation for Directly Loaded Fuel in Normal Conditions of Transport

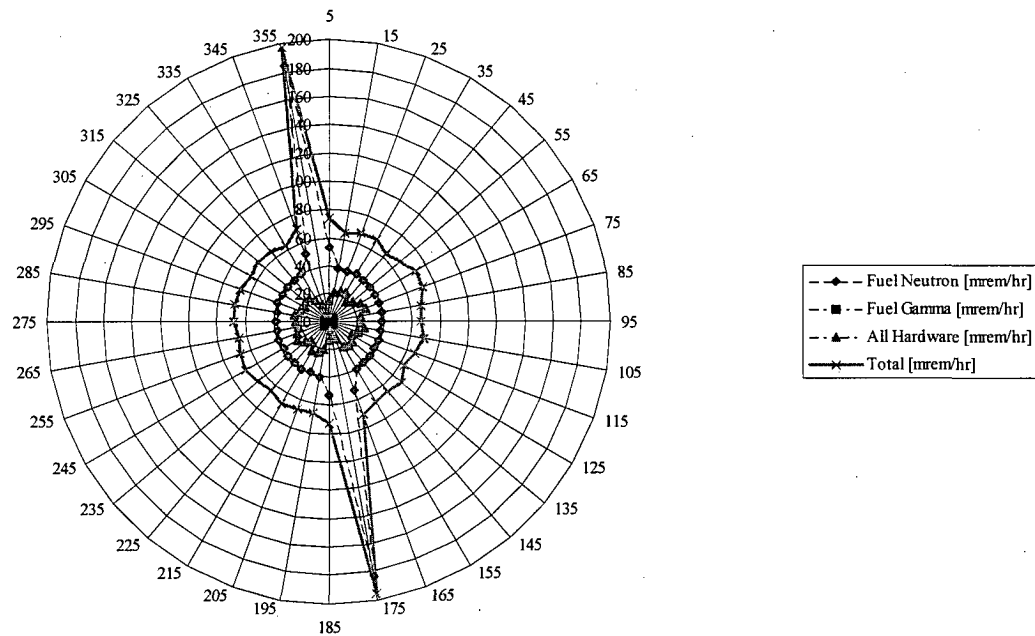


Figure 5.4-4 Azimuthal Radial Surface Dose Rate Profile by Source Type over Heat Fin Axial Extent for Directly Loaded Fuel in Normal Conditions of Transport

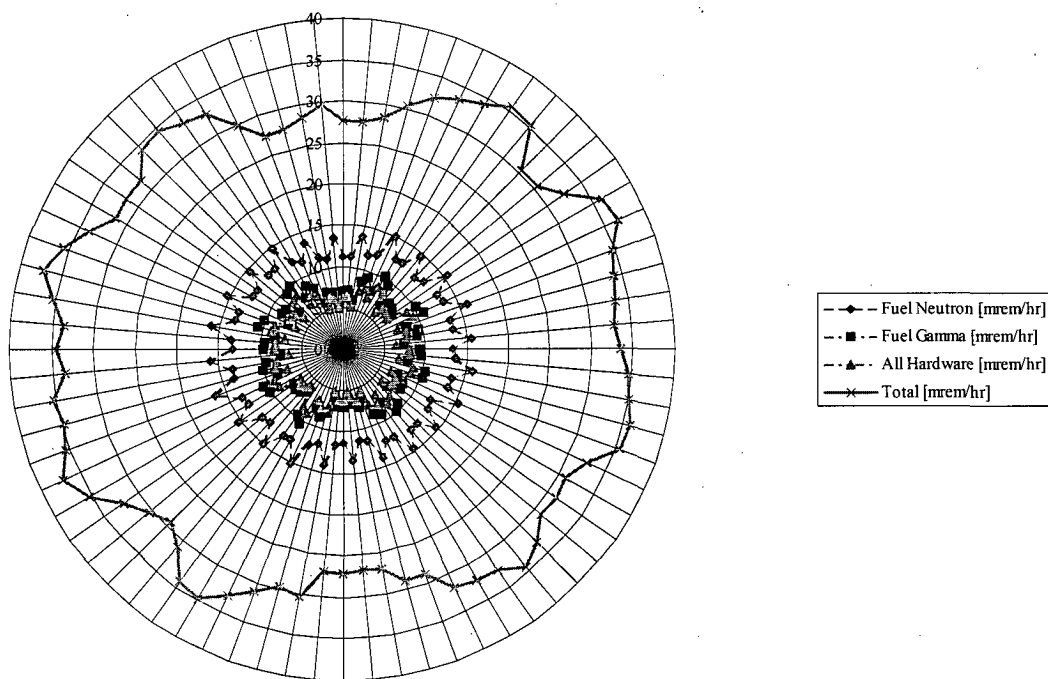


Figure 5.4-5 Radial Dose Rate Profile by Source Type at 1 meter for Directly Loaded Fuel in the Accident Condition

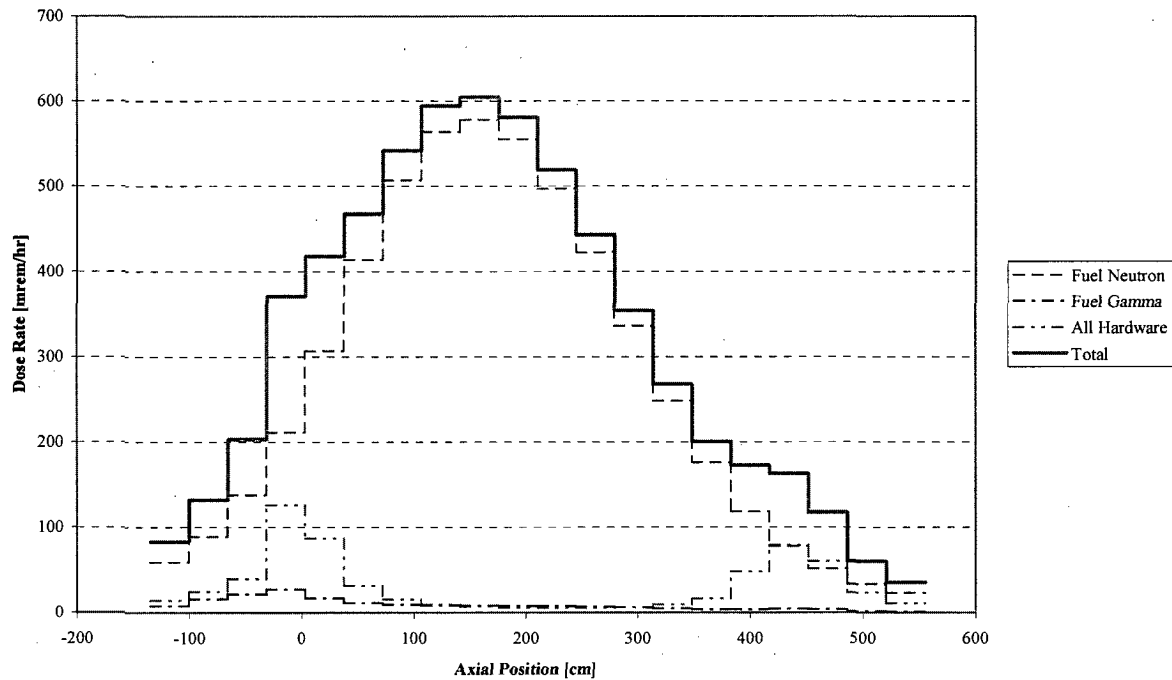


Figure 5.4-6 Azimuthal Radial Dose Rate Profile at 1 meter for Directly Loaded Fuel in the Accident Condition

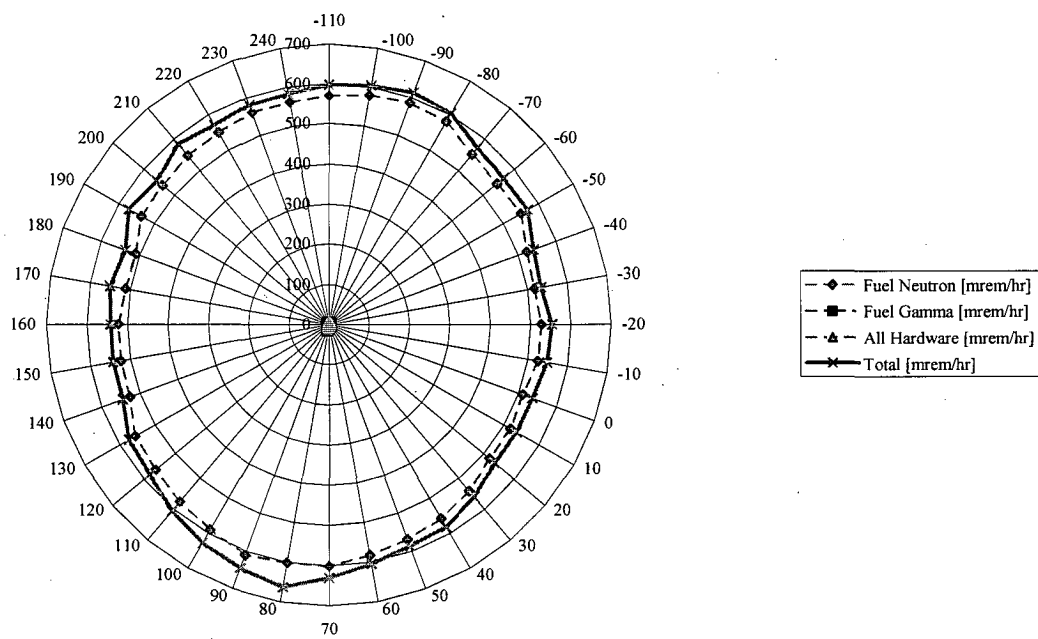


Figure 5.4-7 Graphical Comparison of Normal Conditions Radial 2m+Railcar Dose Rate Profile for DRM and Direct Solution – 14×14 Assembly at 40,000 MWD/MTU, 3.7 wt % ^{235}U , 7 Years Cool Time

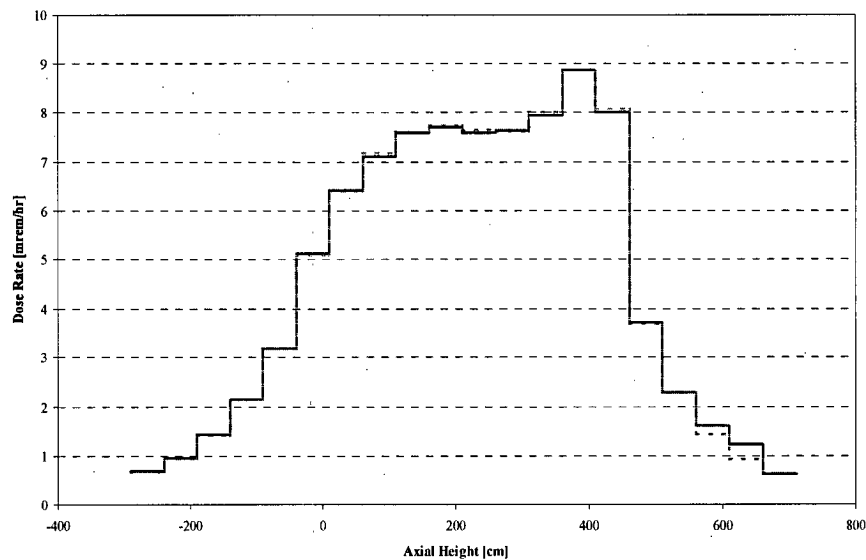


Figure 5.4-8 Graphical Comparison of Accident Conditions Radial 1m Dose Rate Profile for DRM and Direct Solution – 15×15 Assembly at 40,000 MWD/MTU, 3.7 wt % ^{235}U , 7 Years Cool Time

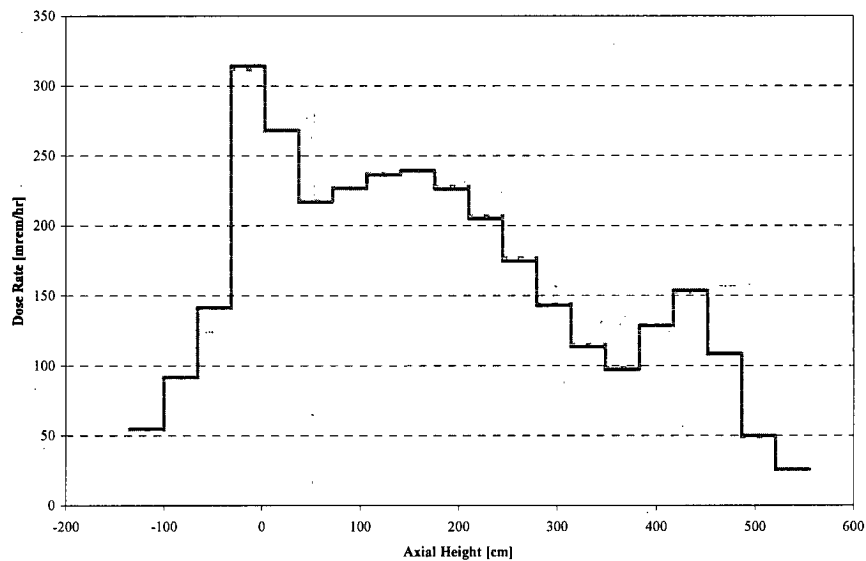
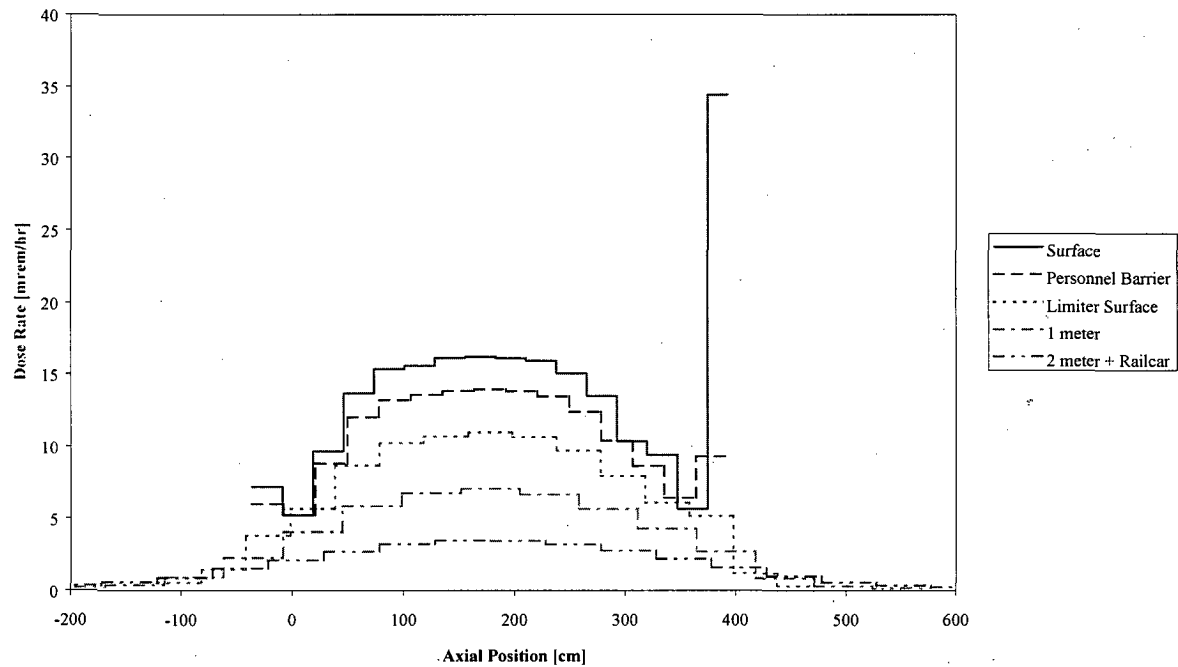
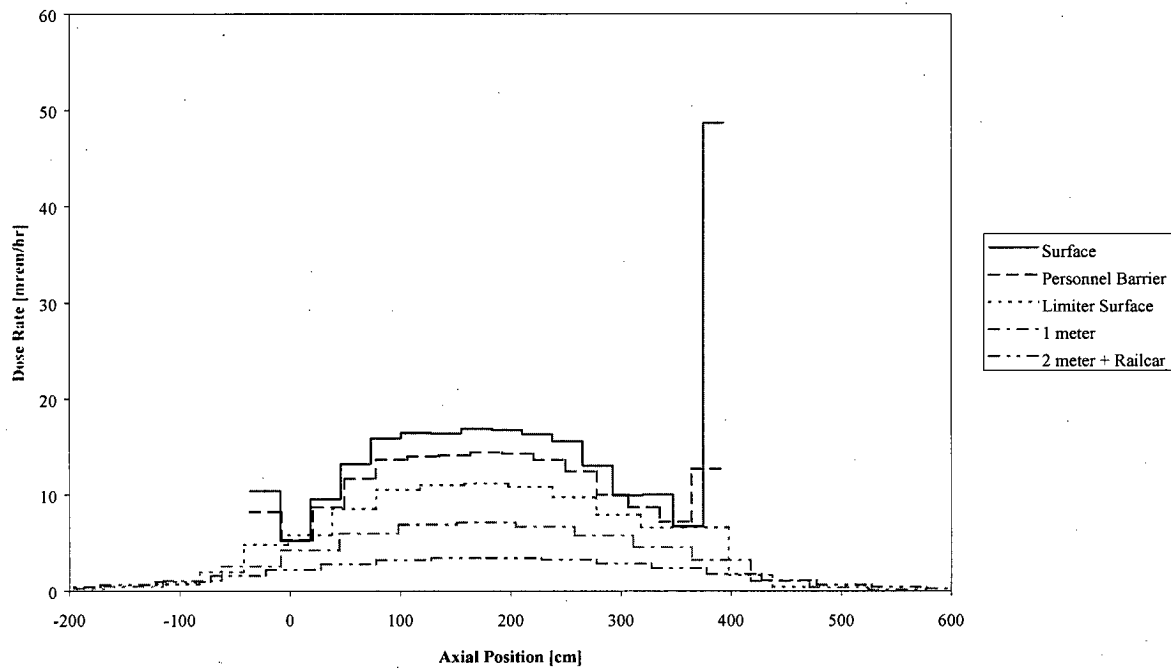


Figure 5.4-9 NAC-STC Radial Dose Rate Profile – Normal Conditions – Design Basis
Connecticut Yankee Stainless Steel Clad Fuel



Note: The surface peak dose rate is due to neutrons streaming through the gap in neutron shielding between the neutron shield and the top impact limiter.

Figure 5.4-10 NAC-STC Radial Dose Rate Profile – Normal Conditions – Design Basis
Connecticut Yankee Zircaloy Clad Fuel



Note: The surface peak dose rate is due to neutrons streaming through the gap in neutron shielding between the neutron shield and the top impact limiter.

Figure 5.4-11 NAC-STC CY-MPC Azimuthal Heat Fin Dose Rate Variations – Normal
Conditions – Design Basis Stainless Steel Clad Fuel

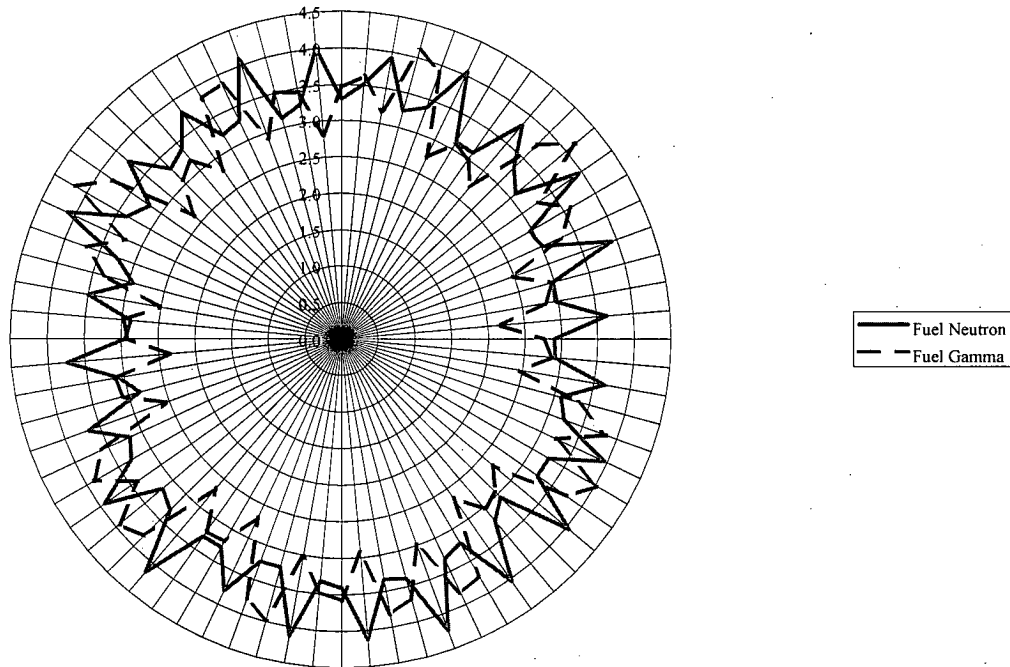


Figure 5.4-12 NAC-STC CY-MPC Azimuthal Heat Fin Dose Rate Variations – Normal
Conditions – Design Basis Zircaloy Clad Fuel

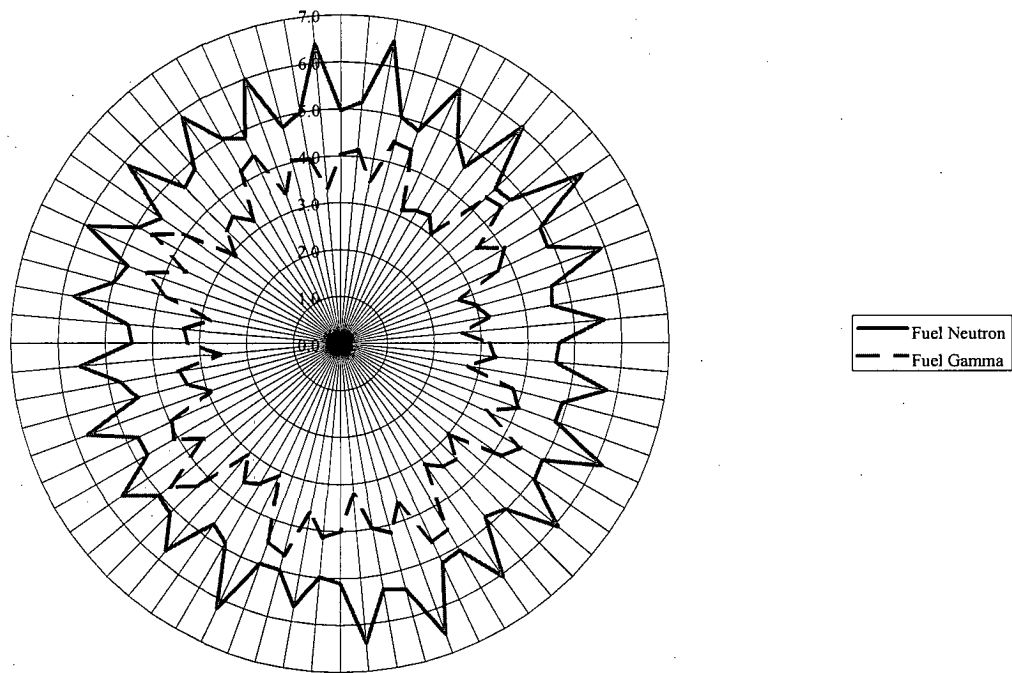


Figure 5.4-13 NAC-STC CY-MPC Radial Dose Rate Profile – Accident Conditions – Design
Basis Stainless Steel Clad Fuel

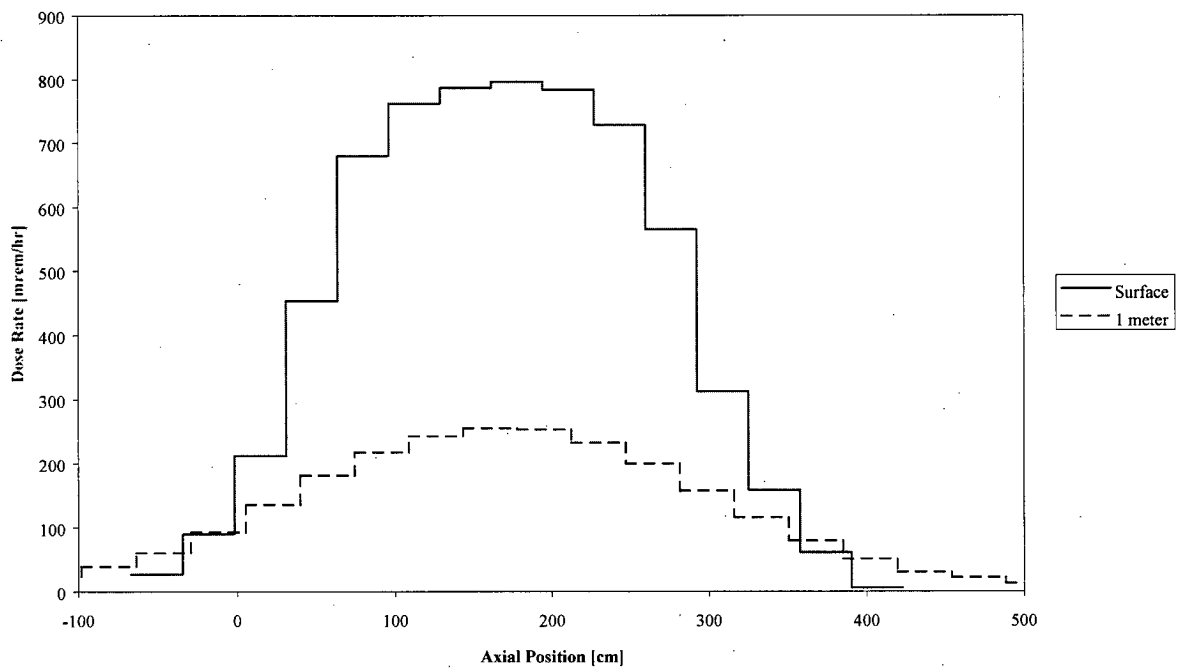


Figure 5.4-14 NAC-STC CY-MPC Radial Dose Rate Profile – Accident Conditions – Design
Basis Zircaloy Clad Fuel

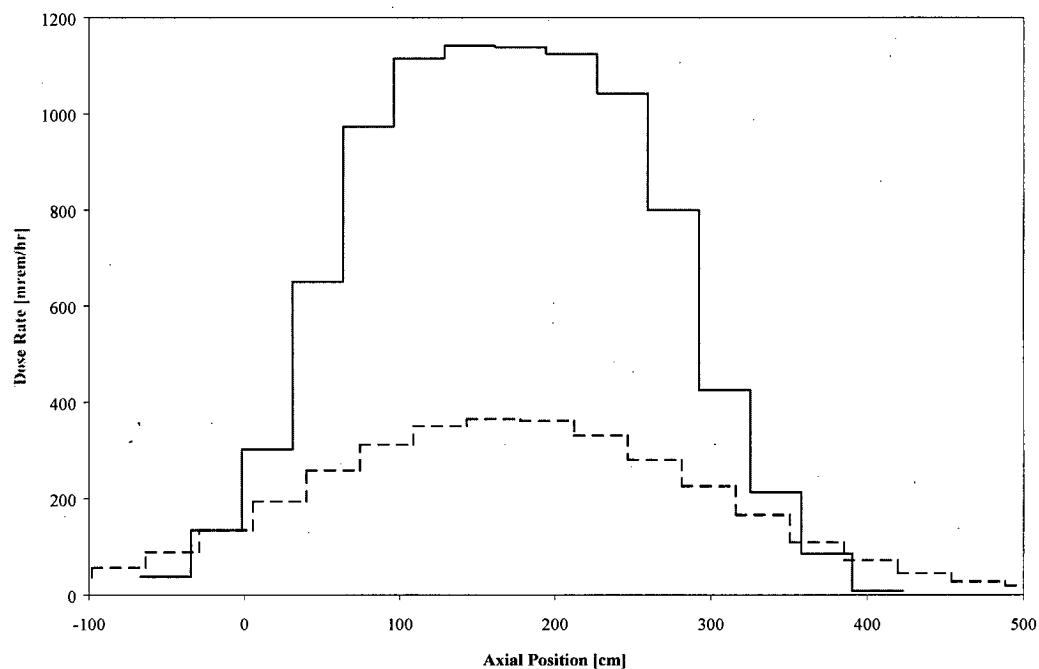


Figure 5.4-15 NAC-STC CY-MPC Radial Dose Rate Profile – Normal Conditions – GTCC
Waste

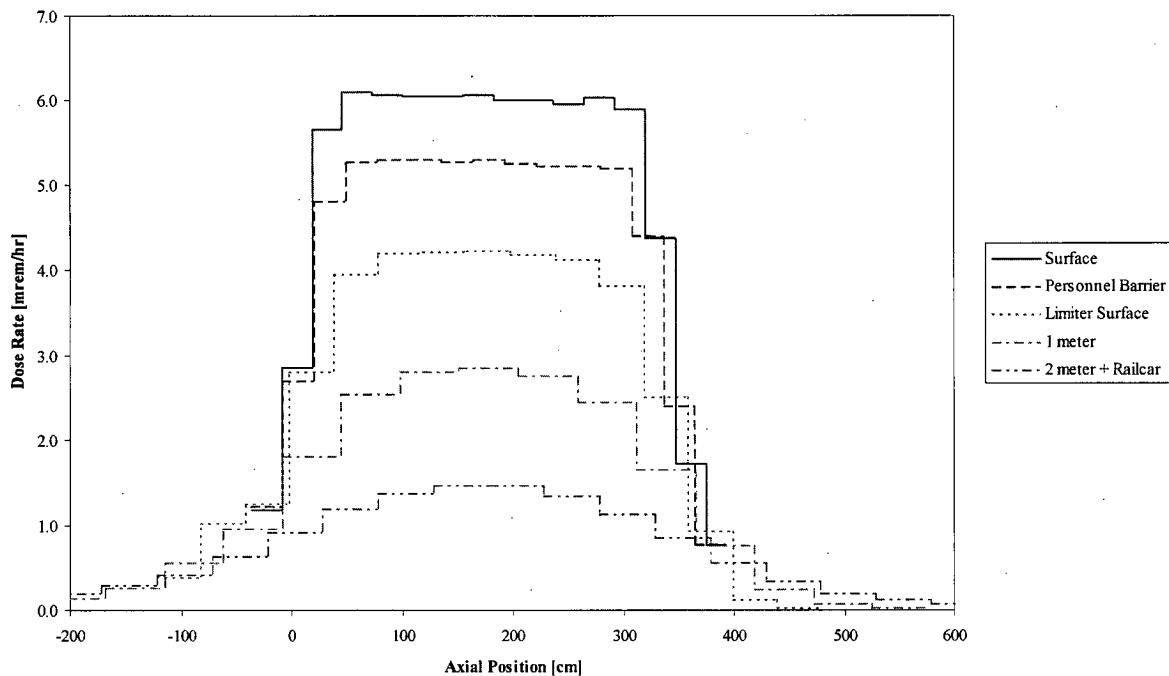
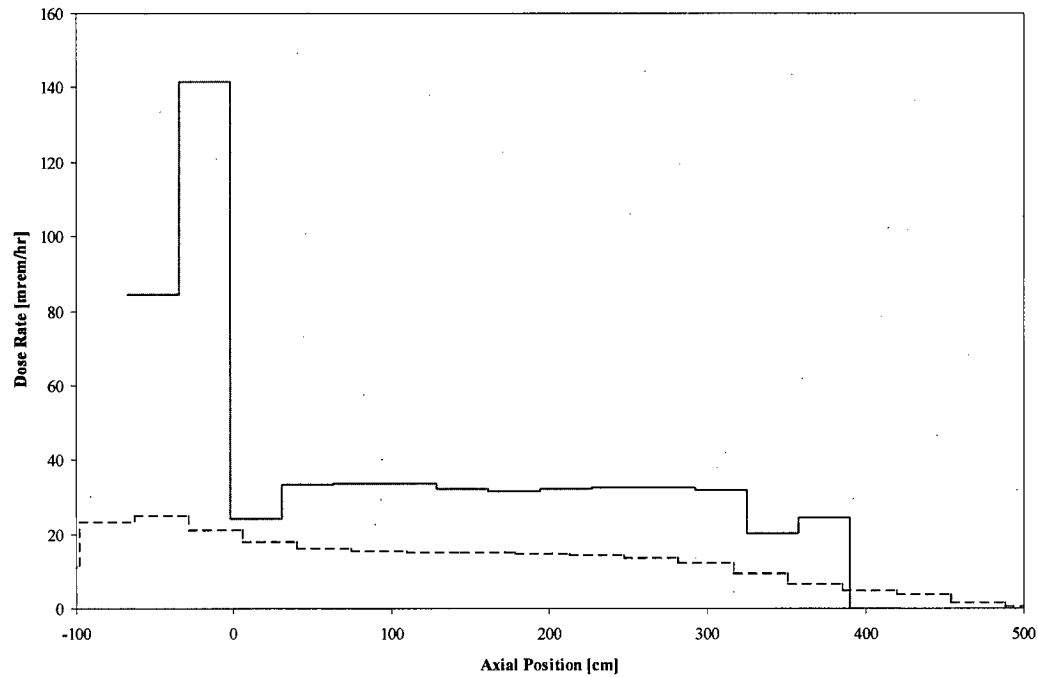


Figure 5.4-16 NAC-STC CY-MPC Radial Dose Rate Profile – Accident Conditions – GTCC Waste



Note: Surface peak is due to gamma streaming through the void created by the modeled bottom axial lead slump and minimal shielding of the canister bottom plate. A similar peak is not observed near the top of the transport cask because of the additional shielding of the canister lids.

Table 5.4-1 ANSI/ANS 6.1.1-1977 Neutron Flux-to-Dose Conversion Factors

Energy [MeV]	Response [(rem/hr)/(n/cm ² /sec)]
20.0	2.27E-04
14.0	2.08E-04
10.0	1.47E-04
7.0	1.47E-04
5.0	1.56E-04
2.5	1.25E-04
1.0	1.32E-04
5.0E-01	9.26E-05
1.0E-01	2.17E-05
1.0E-02	3.56E-06
1.0E-03	3.76E-06
1.0E-04	4.18E-06
1.0E-05	4.54E-06
1.0E-06	4.46E-06
1.0E-07	3.67E-06
2.5E-08	3.67E-06

Table 5.4-2 ANSI/ANS 6.1.1-1977 Gamma Flux-to-Dose Conversion Factors

Energy, E [MeV]	Response [(rem/hr)/(γ/cm ² /sec)]	Energy, E [MeV]	Response [(rem/hr)/(γ/cm ² /sec)]
15.0	1.33E-05	1.0	1.98E-06
13.0	1.18E-05	0.8	1.68E-06
11.0	1.03E-05	0.7	1.52E-06
9.0	8.77E-06	0.65	1.44E-06
7.5	7.66E-06	0.6	1.36E-06
6.75	7.11E-06	0.55	1.27E-06
6.25	6.74E-06	0.5	1.17E-06
5.75	6.37E-06	0.45	1.08E-06
5.25	6.01E-06	0.4	9.85E-07
5.0	5.80E-06	0.35	8.78E-07
4.75	5.60E-06	0.3	7.59E-07
4.25	5.23E-06	0.25	6.31E-07
3.75	4.83E-06	0.2	5.01E-07
3.25	4.41E-06	0.15	3.79E-07
2.8	4.01E-06	0.1	2.83E-07
2.6	3.82E-06	0.07	2.58E-07
2.2	3.42E-06	0.05	2.90E-07
1.8	2.99E-06	0.03	5.82E-07
1.4	2.51E-06	0.01	3.96E-06

Table 5.4-3 Minimum Cooling Time Evaluation for 14×14 Reference Fuel

Enrichment [wt % ²³⁵ U]	Minimum Cooling Time (Years) 40,000 MWD/MTU			Active Constraint
	Decay Heat 850 W/assy	2m+Railcar 9.5 mrem/hr	Limiting	
1.7	-	-	-	-
1.9	6.2	11.6	12	2m+Railcar
2.1	6.1	10.8	11	2m+Railcar
2.3	6.0	10.0	10	2m+Railcar
2.5	5.9	9.3	10	2m+Railcar
2.7	5.8	8.7	9	2m+Railcar
2.9	5.8	8.2	9	2m+Railcar
3.1	5.7	7.7	8	2m+Railcar
3.3	5.6	7.3	8	2m+Railcar
3.5	5.6	6.9	7	2m+Railcar
3.7	5.5	6.5	7	2m+Railcar
3.9	5.5	6.2	7	2m+Railcar
4.1	5.4	5.9	6	2m+Railcar
4.3	5.4	5.7	6	2m+Railcar
4.5	5.3	5.6	6	2m+Railcar

Table 5.4-4 Radial Dose Rate Loading Table Results for Directly Loaded Fuel in Normal Conditions of Transport

Assembly	Source Term			Radial Dose Rate [mrem/hr]	
	Burnup	Enrichment	Cool Time		
	[MWD/MTU]	[wt % ²³⁵ U]	[years]	Surface	2m+Railcar
14×14	40,000	2.3	10	366.3	9.5
15×15	40,000	2.5	9	337.8	9.4
16×16	40,000	1.9	9	317.4	9.2
17×17	40,000	2.3	9	358.9	9.2

Table 5.4-5 Loading Table for Directly Loaded PWR Fuel

Minimum Initial Enrichment wt % ²³⁵ U (E)	Burnup ≤30 GWD/MTU Minimum Cooling Time [years]				30< Burnup ≤35 GWD/MTU Minimum Cooling Time [years]			
	14×14	15×15	16×16	17×17	14×14	15×15	16×16	17×17
1.7 ≤ E < 1.9	8	7	6	7	10	10	7	9
1.9 ≤ E < 2.1	7	7	5	7	9	9	7	8
2.1 ≤ E < 2.3	7	7	5	6	9	8	6	8
2.3 ≤ E < 2.5	6	6	5	6	8	8	6	7
2.5 ≤ E < 2.7	6	6	5	6	8	7	6	7
2.7 ≤ E < 2.9	6	6	5	5	7	7	5	6
2.9 ≤ E < 3.1	6	5	5	5	7	7	5	6
3.1 ≤ E < 3.3	5	5	5	5	7	6	5	6
3.3 ≤ E < 3.5	5	5	5	5	6	6	5	6
3.5 ≤ E < 3.7	5	5	5	5	6	6	5	6
3.7 ≤ E < 3.9	5	5	5	5	6	6	5	6
3.9 ≤ E < 4.1	5	5	5	5	6	6	5	6
4.1 ≤ E < 4.3	5	5	5	5	5	6	5	6
4.3 ≤ E < 4.5	-	-	-	5	-	-	-	6
E ≥ 4.5	-	-	-	5	-	-	-	6
Minimum Initial Enrichment wt % ²³⁵ U (E)	35< Burnup ≤40 GWD/MTU Minimum Cooling Time [years]				40< Burnup ≤45 GWD/MTU Minimum Cooling Time [years]			
	14×14	15×15	16×16	17×17	14×14	15×15	16×16	17×17
1.7 ≤ E < 1.9	-	-	-	-	-	-	-	-
1.9 ≤ E < 2.1	12	13	9	11	-	-	-	-
2.1 ≤ E < 2.3	11	11	8	10	-	-	-	-
2.3 ≤ E < 2.5	10	10	8	9	14	15	12	14
2.5 ≤ E < 2.7	10	9	7	9	13	14	10	12
2.7 ≤ E < 2.9	9	9	7	8	12	12	9	11
2.9 ≤ E < 3.1	9	8	6	8	11	11	8	10
3.1 ≤ E < 3.3	8	8	6	7	10	10	8	9
3.3 ≤ E < 3.5	8	7	6	7	10	10	7	9
3.5 ≤ E < 3.7	7	7	6	7	9	9	7	9
3.7 ≤ E < 3.9	7	7	6	7	9	9	7	9
3.9 ≤ E < 4.1	7	7	6	7	8	9	7	9
4.1 ≤ E < 4.3	6	7	6	7	8	8	7	9
4.3 ≤ E < 4.5	-	-	-	7	-	-	-	8
E ≥ 4.5	-	-	-	7	-	-	-	8

Table 5.4-6 Detector Maximum Dose Rates for Directly Loaded Fuel in Normal Conditions of Transport

Detector	Source	Surface		2 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	0.4	0.2%	0.1	0.2%
	Fuel Gamma	0.2	0.8%	0.0	1.1%
	Fuel Hardware	0.3	1.8%	0.0	2.2%
	Fuel N-Gamma	0.1	1.3%	0.0	1.4%
	Upper Plenum	2.1	0.5%	0.4	0.5%
	Upper Nozzle	3.0	0.6%	0.6	0.6%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	6.1	0.3%	1.3	0.3%
Radial	Fuel Neutron	152.2	0.3%	2.9	0.3%
	Fuel Gamma	2.2	3.8%	1.2	0.8%
	Fuel Hardware	6.1	5.7%	0.5	4.4%
	Fuel N-Gamma	1.4	4.7%	0.8	0.9%
	Upper Plenum	87.2	0.4%	1.8	0.4%
	Upper Nozzle	117.3	0.4%	2.2	0.5%
	Lower Nozzle	0.0	15.0%	0.0	0.8%
	Total	366.3	0.2%	9.5	0.3%
Bottom Axial	Fuel Neutron	4.0	0.3%	0.7	0.2%
	Fuel Gamma	0.7	1.1%	0.1	0.7%
	Fuel Hardware	2.3	0.9%	0.4	4.1%
	Fuel N-Gamma	0.4	1.3%	0.1	0.9%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	7.0	0.8%	1.3	0.7%
	Total	14.3	0.4%	2.6	0.7%

Note: Dose rates at 2 meter locations radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-7 Detector Maximum Dose Rates for Directly Loaded Fuel in Accident Conditions

Detector	Source	Surface		1 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	31.2	0.5%	23.3	0.9%
	Fuel Gamma	1.8	2.6%	1.2	2.0%
	Fuel Hardware	1.5	4.6%	0.9	30.2%
	Fuel N-Gamma	0.5	1.3%	0.4	2.8%
	Upper Plenum	11.8	1.0%	6.2	1.0%
	Upper Nozzle	4.8	1.7%	2.0	2.0%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	51.7	0.5%	34.1	1.1%
Radial	Fuel Neutron	1585.5	0.2%	577.8	0.2%
	Fuel Gamma	15.2	1.5%	8.1	0.9%
	Fuel Hardware	27.8	12.2%	11.8	11.7%
	Fuel N-Gamma	8.3	1.8%	4.3	1.1%
	Upper Plenum	0.0	0.0%	0.1	2.6%
	Upper Nozzle	0.0	0.0%	0.1	5.4%
	Lower Nozzle	0.0	44.5%	2.4	0.6%
	Total	1636.7	0.3%	604.6	0.3%
Bottom Axial	Fuel Neutron	119.3	0.3%	51.2	6.3%
	Fuel Gamma	20.5	1.2%	5.4	1.3%
	Fuel Hardware	5.0	4.7%	1.4	4.4%
	Fuel N-Gamma	5.4	1.1%	1.5	1.2%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	36.3	1.1%	9.6	1.3%
	Total	186.5	0.3%	69.1	4.7%

Note: The azimuthal maximum radial dose rates are 1728.3 (1.8%) and 663.9 (4.4%) mrem/hr at the surface and at 1 meter from the surface, respectively.

Table 5.4-8 Directly Loaded Radial Detector Description for Normal Conditions of Transport

Description	Inner Radius [cm]	Axial				Azimuthal	
		Lower [cm]	Upper [cm]	Divisions	Band [cm]	Divisions	Start Angle
Surface	124.71	-3.81	406.70	15	27.37	1	0
Package	110.11	406.70	425.60	1	18.90	1	0
Heat Fin	125.71	44.38	405.50	1	361.12	72	2.5
Trunnion	126.71	-3.81	19.05	1	22.86	36	5
2m+Railcar	357.48	-289.92	710.69	20	50.03	1	0

Table 5.4-9 Directly Loaded Radial Detector Description for Accident Conditions of Transport

Description	Inner Radius [cm]	Axial				Azimuthal	
		Lower [cm]	Upper [cm]	Divisions	Band [cm]	Divisions	Start Angle
Surface	124.71	-34.67	455.45	15	32.67	1	0
Surface Azi.	125.71	185.68	215.68	1	30.00	36	-110
1m	210.11	-134.67	555.45	20	34.51	1	0
1m Azi.	211.11	185.68	215.68	1	30.00	36	-110

Table 5.4-10 CY-MPC Neutron Flux-to-Dose Conversion Factors

Group	Upper E [MeV]	Lower E [MeV]	Response [(mrem/hr)/(n/cm ² /sec)]
1	1.46E+01	1.36E+01	2.0533E-01
2	1.36E+01	1.25E+01	1.8999E-01
3	1.25E+01	1.13E+01	1.7250E-01
4	1.13E+01	1.00E+01	1.5399E-01
5	1.00E+01	8.25E+00	1.4700E-01
6	8.25E+00	7.00E+00	1.4700E-01
7	7.00E+00	6.07E+00	1.4929E-01
8	6.07E+00	4.72E+00	1.5348E-01
9	4.72E+00	3.68E+00	1.4580E-01
10	3.68E+00	2.87E+00	1.3478E-01
11	2.87E+00	1.74E+00	1.2657E-01
12	1.74E+00	6.40E-01	1.2570E-01
13	6.40E-01	3.90E-01	8.8205E-02
14	3.90E-01	1.10E-01	4.6004E-02
15	1.10E-01	6.74E-02	1.8108E-02
16	6.74E-02	2.48E-02	1.0774E-02
17	2.48E-02	9.12E-03	4.9057E-03
18	9.12E-03	2.95E-03	3.6168E-03
19	2.95E-03	9.61E-04	3.7152E-03
20	9.61E-04	3.54E-04	3.8611E-03
21	3.54E-04	1.66E-04	4.0252E-03
22	1.66E-04	4.81E-05	4.1919E-03
23	4.81E-05	1.60E-05	4.3795E-03
24	1.60E-05	4.00E-06	4.5200E-03
25	4.00E-06	1.50E-06	4.4895E-03
26	1.50E-06	5.50E-07	4.3924E-03
27	5.50E-07	7.09E-08	3.9685E-03
28	7.09E-08	0.00E+00	2.3759E-03

Table 5.4-11 CY-MPC Gamma Flux-to-Dose Conversion Factors

Group	Upper E [MeV]	Lower E [MeV]	Response [(mrem/hr)/(γ/cm ² /sec)]
1	1.40E+01	1.20E+01	1.1728E-02
2	1.20E+01	1.00E+01	1.0225E-02
3	1.00E+01	8.00E+00	8.7164E-03
4	8.00E+00	6.50E+00	7.4457E-03
5	6.50E+00	5.00E+00	6.3551E-03
6	5.00E+00	4.00E+00	5.3991E-03
7	4.00E+00	3.00E+00	4.5984E-03
8	3.00E+00	2.50E+00	3.9449E-03
9	2.50E+00	2.00E+00	3.4485E-03
10	2.00E+00	1.66E+00	2.9982E-03
11	1.66E+00	1.44E+00	2.6706E-03
12	1.44E+00	1.22E+00	2.3929E-03
13	1.22E+00	1.00E+00	2.1055E-03
14	1.00E+00	8.00E-01	1.8164E-03
15	8.00E-01	6.00E-01	1.5143E-03
16	6.00E-01	4.00E-01	1.1686E-03
17	4.00E-01	3.00E-01	8.6947E-04
18	3.00E-01	2.00E-01	6.2398E-04
19	2.00E-01	1.00E-01	3.8050E-04
20	1.00E-01	5.00E-02	2.7163E-04
21	5.00E-02	2.00E-02	5.8620E-04
22	2.00E-02	1.00E-02	2.3540E-03

Table 5.4-12 NAC-STC CY-MPC Detector Maximum Dose Rates – Normal Conditions –
Design Basis Stainless Steel Clad Fuel

Detector	Source	Surface		2 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	0.1	0.5%	0.1	2.8%
	Fuel Gamma	0.0	4.5%	0.0	7.1%
	Fuel Hardware	0.0	3.4%	0.0	4.6%
	Fuel N-Gamma	0.1	5.3%	0.0	7.3%
	Upper Plenum	0.0	1.9%	0.0	2.0%
	Upper Nozzle	0.0	1.0%	0.0	0.9%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	0.3	2.4%	0.1	2.1%
Radial	Fuel Neutron	30.2	0.5%	0.9	0.6%
	Fuel Gamma	0.0	14.0%	0.9	0.7%
	Fuel Hardware	0.1	16.6%	1.5	0.8%
	Fuel N-Gamma	1.4	5.9%	0.2	14.9%
	Upper Plenum	0.3	1.6%	0.0	0.5%
	Upper Nozzle	2.1	0.6%	0.0	0.4%
	Lower Nozzle	0.0	58.0%	0.0	0.9%
	Total	34.0	0.5%	3.6	1.0%
Bottom Axial	Fuel Neutron	0.5	0.4%	0.1	1.2%
	Fuel Gamma	0.1	1.0%	0.0	0.9%
	Fuel Hardware	0.1	1.2%	0.0	1.0%
	Fuel N-Gamma	0.3	1.6%	0.0	1.6%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	0.6	0.6%	0.1	0.5%
	Total	1.6	0.4%	0.3	0.5%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-13 NAC-STC CY-MPC Detector Maximum Dose Rates – Normal Conditions –
Design Basis Zircaloy Clad Fuel

Detector	Source	Surface		2 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	0.2	0.7%	0.1	3.7%
	Fuel Gamma	0.0	6.5%	0.0	5.0%
	Fuel Hardware	0.0	7.7%	0.0	3.9%
	Fuel N-Gamma	0.2	2.1%	0.0	3.9%
	Upper Plenum	0.0	2.1%	0.0	2.4%
	Upper Nozzle	0.0	0.6%	0.0	0.6%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	0.4	1.0%	0.2	2.2%
Radial	Fuel Neutron	43.6	0.4%	1.3	0.5%
	Fuel Gamma	0.0	11.4%	1.0	0.9%
	Fuel Hardware	0.0	18.1%	0.8	0.7%
	Fuel N-Gamma	2.8	10.9%	0.3	7.5%
	Upper Plenum	0.3	1.6%	0.0	0.4%
	Upper Nozzle	2.4	0.6%	0.1	0.3%
	Lower Nozzle	0.0	100.0%	0.0	0.8%
	Total	49.1	0.7%	3.6	0.8%
Bottom Axial	Fuel Neutron	0.8	0.4%	0.1	0.9%
	Fuel Gamma	0.1	0.8%	0.0	0.7%
	Fuel Hardware	0.1	1.1%	0.0	0.9%
	Fuel N-Gamma	0.6	6.2%	0.1	4.4%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	0.4	0.6%	0.1	0.5%
	Total	2.0	1.8%	0.3	1.1%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-14 NAC-STC CY-MPC Detector Maximum Dose Rates – Accident Conditions –
Design Basis Stainless Steel Clad Fuel

Detector	Source	Surface		1 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	2.1	0.6%	9.5	1.7%
	Fuel Gamma	0.0	5.1%	0.1	30.0%
	Fuel Hardware	0.0	2.6%	0.1	12.8%
	Fuel N-Gamma	0.2	11.8%	0.0	7.0%
	Upper Plenum	0.0	7.1%	0.2	5.7%
	Upper Nozzle	0.0	6.5%	0.8	0.9%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	2.4	1.3%	10.8	1.5%
Radial	Fuel Neutron	746	0.6%	234	0.4%
	Fuel Gamma	21	5.4%	9	2.4%
	Fuel Hardware	31	2.8%	14	1.3%
	Fuel N-Gamma	9	59.3%	1	15.2%
	Upper Plenum	0	0.0%	0	2.1%
	Upper Nozzle	0	0.0%	0	1.2%
	Lower Nozzle	0	0.0%	0	3.6%
	Total	806	0.9%	259	0.4%
Bottom Axial	Fuel Neutron	4.7	0.5%	12.3	3.2%
	Fuel Gamma	0.3	1.0%	0.0	12.8%
	Fuel Hardware	0.4	1.1%	0.1	14.3%
	Fuel N-Gamma	1.1	1.9%	0.1	32.3%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	2.0	0.7%	0.5	12.0%
	Total	8.5	0.4%	13.0	3.1%

Table 5.4-15 NAC-STC CY-MPC Detector Maximum Dose Rates – Accident Conditions –
Design Basis Zircaloy Clad Fuel

Detector	Source	Surface		1 meter	
		mrem/hr	RSD	mrem/hr	RSD
Top Axial	Fuel Neutron	3.1	0.6%	14.1	1.7%
	Fuel Gamma	0.0	2.1%	0.1	10.9%
	Fuel Hardware	0.0	4.4%	0.1	18.5%
	Fuel N-Gamma	0.3	5.4%	0.1	6.3%
	Upper Plenum	0.0	4.6%	0.2	2.6%
	Upper Nozzle	0.0	4.6%	1.0	1.0%
	Lower Nozzle	0.0	0.0%	0.0	0.0%
	Total	3.4	0.8%	15.4	1.6%
Radial	Fuel Neutron	1,123	0.6%	348	0.4%
	Fuel Gamma	23	3.3%	10	1.6%
	Fuel Hardware	19	3.2%	8	1.6%
	Fuel N-Gamma	5	31.0%	2	13.7%
	Upper Plenum	0	0.0%	0	2.1%
	Upper Nozzle	0	0.0%	0	1.7%
	Lower Nozzle	0	0.0%	0	7.1%
	Total	1,170	0.6%	369	0.4%
Bottom Axial	Fuel Neutron	7.6	0.5%	17.6	3.4%
	Fuel Gamma	0.4	1.1%	0.2	43.1%
	Fuel Hardware	0.3	0.8%	0.1	12.3%
	Fuel N-Gamma	1.7	3.9%	0.3	25.8%
	Upper Plenum	0.0	0.0%	0.0	0.0%
	Upper Nozzle	0.0	0.0%	0.0	0.0%
	Lower Nozzle	1.3	0.7%	0.4	12.1%
	Total	11.3	0.7%	18.5	3.3%

Table 5.4-16 NAC-STC CY-MPC Detector Average Dose Rates – Normal Conditions – Design Basis Stainless Steel Clad Fuel

Detector	Surface		2 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	0.2	2.3%	0.1	3.4%
Radial	14.3	6.3%	1.6	5.3%
Bottom Axial	0.7	1.5%	0.2	3.2%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-17 NAC-STC CY-MPC Detector Average Dose Rates – Normal Conditions – Design Basis Zircaloy Clad Fuel

Detector	Surface		2 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	0.3	1.4%	0.1	2.8%
Radial	15.5	1.2%	1.7	1.0%
Bottom Axial	1.0	2.0%	0.2	2.7%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-18 NAC-STC CY-MPC Detector Average Dose Rates – Accident Conditions –
Design Basis Stainless Steel Clad Fuel

Detector	Surface		1 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	1.5	1.0%	4.3	2.3%
Radial	444	0.8%	125	0.6%
Bottom Axial	5.0	0.6%	5.8	5.9%

Table 5.4-19 NAC-STC CY-MPC Detector Average Dose Rates – Accident Conditions –
Design Basis Zircaloy Clad Fuel

Detector	Surface		1 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	2.1	0.9%	6.2	2.3%
Radial	636	0.6%	178	0.6%
Bottom Axial	6.9	1.0%	8.3	4.5%

Table 5.4-20 NAC-STC CY-MPC Detector Maximum Dose Rates – Normal Conditions –
Design Basis GTCC Waste

Detector	Surface		2 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	<0.1	1.2%	<0.1	1.4%
Radial	6.1	0.6%	1.5	0.3%
Bottom Axial	4.2	0.2%	0.7	0.2%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-21 NAC-STC CY-MPC Detector Maximum Dose Rates – Accident Conditions –
Design Basis GTCC Waste

Detector	Surface		1 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	<0.1	8.5%	0.4	11.4%
Radial	141.4	10.8%	25.0	5.7%
Bottom Axial	16.7	0.2%	5.2	0.2%

Table 5.4-22 NAC-STC CY-MPC Detector Average Dose Rates – Normal Conditions – Design Basis GTCC Waste

Detector	Surface		2 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	<0.1	2.2%	<0.1	4.8%
Radial	4.8	0.7%	0.6	0.4%
Bottom Axial	1.1	0.3%	0.2	1.0%

Note: Dose rates at 2 meter location radially are 2 meters from the railcar. Dose rates at 2 meter locations axially are measured from the ends of the impact limiters.

Table 5.4-23 NAC-STC CY-MPC Detector Average Dose Rates – Accident Conditions – Design Basis GTCC Waste

Detector	Surface		1 meter	
	mrem/hr	RSD	mrem/hr	RSD
Top Axial	<0.1	12.7%	0.1	13.5%
Radial	39.2	10.2%	12.5	5.0%
Bottom Axial	5.9	0.4%	2.3	3.5%

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5.5 Sample Input Files

This section contains sample input files employed in the shielding evaluations of the NAC-STC.

5.5.1 Sample Input Files for Directly Loaded Fuel

Sample input files for directly loaded fuel are provided in Figures 5.5-1 through 5.5-4.

Figure 5.5-1 SAS2H Input File for Directly Loaded 14×14 Fuel at 40,000 MWD/MTU and 2.3 wt % ²³⁵U

```
=SAS2H      PARM=(HALT09,SKIPSHIPDATA)
Class 1 - aa14b - STC Hybrid14 (Rev 0) - 2.3 w/o U235, 40000 MWD/MTU, 5 - 16 years cool time
27GROUPNDF4 LATTICECELL
UO2      1 0.950 900 92235 2.3 92238 97.7 END
ZIRCALLOY 2 1.0 620 END
H2O      3 DEN=0.725 1.0 580 END
ARBM-BORMOD 0.725 1 1 0 0 5000 100 3 550.0E-6 580 END
ZIRCALLOY 4 1.0 580 END
H2O      5 DEN=0.725 0.9476 580 END
ZIRCALLOY 5 0.0524 580 END
END COMP
SQUAREPITCH 1.4122 0.9332 1 3 1.0719 2 0.9576 0 END
NPIN=179 FUEL=368.808 NCYC=3 NLIB=3 PRIN=6 LIGH=5
INPL=1 NUMH=16 NUMI=1 MXTUBE=4 ORTU=0.6102 SRTU=0.5239 END
POWER=13.7183 BURN=402.7954 DOWN=60 END
POWER=13.7183 BURN=402.7954 DOWN=60 END
POWER=13.7183 BURN=402.7954 DOWN=1461 END
FE 0.6738 CR 0.1900 NI 0.1150 MN 0.0200 CO 0.0012
END
=ORIGENS
0$$$ A4 21 A8 26 A10 51 71 E
1$$$ 1 1T
COOLING 5 - 16 YEARS AND FISSION PRODUCT GAMMA REBIN
3$$$ 21 0 1 28 A33 22 E
54$$$ A8 1 E T
35$$$ 0 T
56$$$ 0 9 A13 -2 5 3 E
57** 4.0 E T
COOLING 5 - 16 YEARS AND FISSION PRODUCT GAMMA REBIN
SINGLE REACTOR ASSEMBLY
60** 5.0 6.0 7.0 8.0 9.0 10.0 12.0 14.0 16.0
65$$$ A4 1 A7 1 A10 1 A25 1 A28 1 A31 1 A46 1 A49 1 A52 1 E
61** F.00000001
81$$$ 2 51 26 1 E
82$$$ F6
83** 1.40e+7 1.20e+7 1.00e+7 8.00e+6 6.50e+6 5.00e+6
      4.00e+6 3.00e+6 2.50e+6 2.00e+6 1.66e+6 1.44e+6
      1.22e+6 1.00e+6 0.80e+6 0.60e+6 0.40e+6 0.30e+6
      0.20e+6 0.10e+6 0.05e+6 0.02e+6 0.01e+6
84** 1.46e+7 1.36e+7 1.25e+7 1.125e+7 1.00e+7
      8.25e+6 7.00e+6 6.07e+6 4.72e+6 3.68e+6
      2.87e+6 1.74e+6 0.64e+6 0.39e+6 0.11e+6
      6.74e+4 2.48e+4 9.12e+3 2.95e+3 9.61e+2
      3.54e+2 1.66e+2 4.81e+1 1.60e+1 4.00e+0
      1.50e+0 5.50e-1 7.09e-2 1.00e-5 T
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
FISSION PRODUCT GAMMA SPECTRA IN AEA GROUPS
56$$$ F0 T
END
=ORIGENS
0$$$ A4 21 A8 26 A10 51 71 E
1$$$ 1 1T
COOLING 5 - 16 YEARS AND ACTINIDE GAMMA REBIN
3$$$ 21 0 1 28 A33 22 E
54$$$ A8 1 E T
35$$$ 0 T
56$$$ 0 9 A13 -2 5 3 E
57** 4.0 E T
COOLING 5 - 16 YEARS AND ACTINIDE GAMMA REBIN
SINGLE REACTOR ASSEMBLY
60** 5.0 6.0 7.0 8.0 9.0 10.0 12.0 14.0 16.0
65$$$ A4 1 A7 1 A10 1 A25 1 A28 1 A31 1 A46 1 A49 1 A52 1 E
61** F.00000001
```


Figure 5.5-1 SAS2H Input File for Directly Loaded 14×14 Fuel at 40,000 MWD/MTU and 2.3
wt % ²³⁵U (continued)

```
81$$ 2 51 26 1 E
82$$ F5
83** 1.40e+7 1.20e+7 1.00e+7 8.00e+6 6.50e+6 5.00e+6
      4.00e+6 3.00e+6 2.50e+6 2.00e+6 1.66e+6 1.44e+6
      1.22e+6 1.00e+6 0.80e+6 0.60e+6 0.40e+6 0.30e+6
      0.20e+6 0.10e+6 0.05e+6 0.02e+6 0.01e+6
84** 1.46e+7 1.36e+7 1.25e+7 1.125e+7 1.00e+7
      8.25e+6 7.00e+6 6.07e+6 4.72e+6 3.68e+6
      2.87e+6 1.74e+6 0.64e+6 0.39e+6 0.11e+6
      6.74e+4 2.48e+4 9.12e+3 2.95e+3 9.61e+2
      3.54e+2 1.66e+2 4.81e+1 1.60e+1 4.00e+0
      1.50e+0 5.50e-1 7.09e-2 1.00e-5 T
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
ACTINIDE GAMMA SPECTRA IN AEA GROUPS
56$$ F0 T
END
-ORIGENS
0$$ A4 21 A8 26 A10 51 71 E
1$$ 1 1T
COOLING 5 - 16 YEARS AND LIGHT ELEMENT GAMMA REBIN
3$$ 21 0 1 28 A33 22 E
54$$ A8 1 E T
35$$ 0 T
56$$ 0 9 A13 -2 5 3 E
57** 4.0 E T
COOLING 5 - 16 YEARS AND LIGHT ELEMENT GAMMA REBIN
SINGLE REACTOR ASSEMBLY
60** 5.0 6.0 7.0 8.0 9.0 10.0 12.0 14.0 16.0
65$$ A4 1 A7 1 A10 1 A25 1 A28 1 A31 1 A46 1 A49 1 A52 1 E
61** F.00000001
81$$ 2 51 26 1 E
82$$ F4
83** 1.40e+7 1.20e+7 1.00e+7 8.00e+6 6.50e+6 5.00e+6
      4.00e+6 3.00e+6 2.50e+6 2.00e+6 1.66e+6 1.44e+6
      1.22e+6 1.00e+6 0.80e+6 0.60e+6 0.40e+6 0.30e+6
      0.20e+6 0.10e+6 0.05e+6 0.02e+6 0.01e+6
84** 1.46e+7 1.36e+7 1.25e+7 1.125e+7 1.00e+7
      8.25e+6 7.00e+6 6.07e+6 4.72e+6 3.68e+6
      2.87e+6 1.74e+6 0.64e+6 0.39e+6 0.11e+6
      6.74e+4 2.48e+4 9.12e+3 2.95e+3 9.61e+2
      3.54e+2 1.66e+2 4.81e+1 1.60e+1 4.00e+0
      1.50e+0 5.50e-1 7.09e-2 1.00e-5 T
LIGHT ELEMENT AEA GROUP STRUCTURE
LIGHT ELEMENT AEA GROUP STRUCTURE
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LIGHT ELEMENT AEA GROUP STRUCTURE
LIGHT ELEMENT AEA GROUP STRUCTURE
LIGHT ELEMENT AEA GROUP STRUCTURE
LIGHT ELEMENT AEA GROUP STRUCTURE
56$$ F0 T
END
```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from Energy Group 7 – Normal Conditions

```

columns 1 200
*
* NAC-STC - aal4b_07g - Fuel Gamma - Radial
* Dry Cavity Conditions
* Normal Transport Conditions
* Transport Model Revision v1.5.1.0
* Shielding Revision v1.5
* STC Source Profile
* Cobalt Concentration of 1.2 g/kg
* Fuel Assembly Shift = Cavity, Basket Shift = None
*
* Parameters
*
@samps = 10000000
*
* Unit 1 Control Data
*
begin control data
  run
  sample limit @samps
  time limit 1000m
  seeds 33113 94930
  chime every [@samps/10] samples
  report interim results
  sbd 30s
  dump intervals 1
end
*
* Unit 3 Output Control
*
*begin output control
* suppress inflows
*end
*
* Unit 4 Material Geometry
*
begin material geometry
* Fuel Assembly Type A - Class 1 - aal4b - STC Hybrid14 (Rev 0)
PART 1 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly Type B - Class 1 - aal4b - STC Hybrid14 (Rev 0)
PART 2 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly in Tube (Type A) v1.1
PART 3
BOX 1 1.4135 1.4135 9.9060 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of basket cavity
ZONES
/Fuel Assembly/ P1 +1
/Space in Tube/ H5 +2 -1
/Fuel Tube/ M8 +3 -2
/Container/ H5 +4 -3 -2 -1
VOLUMES UNITY
* Fuel Assembly in Tube (Type B) v1.1
PART 4

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```

BOX 1 1.4135 1.4135 9.9060 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of basket cavity
ZONES
/Fuel Assembly/ P2 +1
/Space in Tube/ H5 +2 -1
/Fuel Tube/ M8 +3 -2
/Container/ H5 +4 -3 -2 -1
VOLUMES UNITY
* Type A Disk Opening with Tube v1.2
PART 5 CLUSTER
BOX P3 0.4547 0.4547 0.0000 22.5450 22.5450 419.1000 ! Fuel tube type A with fuel assy
BOX H5 0.0000 0.0000 0.0000 23.4544 23.4544 419.1000 ! Support disk opening width
* Type B Disk Opening with Tube v1.2
PART 6 CLUSTER
BOX P4 0.4547 0.4547 0.0000 22.5450 22.5450 419.1000 ! Fuel tube type B with fuel assy
BOX H5 0.0000 0.0000 0.0000 23.4544 23.4544 419.1000 ! Support disk opening width
* STC Basket v1.2
PART 7
BOX 1 -38.9153 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 1
BOX 2 -11.7272 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 2
BOX 3 15.4610 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 3
BOX 4 -70.7136 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 4
BOX 5 -38.9153 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 5
BOX 6 -11.7272 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 6
BOX 7 15.4610 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 7
BOX 8 47.2592 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 8
BOX 9 -70.7136 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 9
BOX 10 -38.9153 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 10
BOX 11 -11.7272 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 11
BOX 12 15.4610 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 12
BOX 13 47.2592 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 13
BOX 14 -70.7136 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 14
BOX 15 -38.9153 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 15
BOX 16 -11.7272 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 16
BOX 17 15.4610 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 17
BOX 18 47.2592 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 18
BOX 19 -70.7136 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 19
BOX 20 -38.9153 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 20
BOX 21 -11.7272 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 21
BOX 22 15.4610 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 22
BOX 23 47.2592 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 23
BOX 24 -38.9153 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 24
BOX 25 -11.7272 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 25
BOX 26 15.4610 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 26
ZROD 27 0.0000 0.0000 0.0000 89.9922 419.1000 ! Basket stack to cavity height
ZONES
/Opening01/ P5 +1
/Opening02/ P5 +2
/Opening03/ P5 +3
/Opening04/ P6 +4
/Opening05/ P5 +5
/Opening06/ P5 +6
/Opening07/ P5 +7
/Opening08/ P6 +8
/Opening09/ P5 +9
/Opening10/ P5 +10
/Opening11/ P5 +11
/Opening12/ P5 +12
/Opening13/ P5 +13
/Opening14/ P5 +14
/Opening15/ P5 +15
/Opening16/ P5 +16
/Opening17/ P5 +17
/Opening18/ P5 +18
/Opening19/ P6 +19
/Opening20/ P5 +20
/Opening21/ P5 +21
/Opening22/ P5 +22
/Opening23/ P6 +23
/Opening24/ P5 +24
/Opening25/ P5 +25

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```

/Opening26/ P5 +26
/Basket/ H1 +27 -1 -2 -3 -4 -5
          -6 -7 -8 -9 -10 -11
          -12 -13 -14 -15 -16 -17
          -18 -19 -20 -21 -22 -23
          -24 -25 -26

VOLUMES UNITY
* Basket in Cask Cavity v1.2
PART 8 NEST
ZROD P7 0.0000 0.0000 0.0000 89.9922 419.1000 ! Basket inserted - Includes gap to lid
ZROD H5 0.0000 0.0000 0.0000 90.1700 419.1000 ! Inserts flood matl to id of stc
* Transport Cask Inner Lid - With Ports v1.5.1.0
PART 9
ZROD 1 0.0000 0.0000 0.0000 100.3300 18.0848 ! Inner lid base
ZROD 2 0.0000 0.0000 18.0848 92.5957 4.7752 ! Inner lid cap
ZROD 3 0.0000 77.9907 0.0000 8.2931 22.8600 ! Drain port
ZROD 4 0.0000 -77.9907 0.0000 8.2931 22.8600 ! Vent port
ZROD 5 0.0000 0.0000 15.2400 85.6234 5.0800 ! Neutron shield
ZROD 6 0.0000 77.9907 15.2400 10.1600 5.0800 ! Cut circle 1 for neutron shield
ZROD 7 0.0000 -77.9907 15.2400 10.1600 5.0800 ! Cut circle 2 for neutron shield
BOX 8 -10.1600 77.9907 15.2400 20.3200 7.6327 5.0800 ! Cut box 1 for neutron shield
BOX 9 -10.1600 -85.6234 15.2400 20.3200 7.6327 5.0800 ! Cut box 2 for neutron shield
ZROD 10 0.0000 0.0000 0.0000 100.3300 22.8600 ! Container

ZONES
/Container/ M0 +10 -1 -2
/LidBase1/ M16 +1 -3 -4 -5 -6 -7
          -8 -9
/LidBase2/ M16 +1 +8 -6
/LidBase3/ M16 +1 +9 -7
/LidBase4/ M16 +1 +6 -3
/LidBase5/ M16 +1 +7 -4
/Nshield/ M15 +5 -6 -7 -8 -9
/LidCap1/ M16 +2 -3 -4 -5 -6 -7
          -8 -9
/LidCap2/ M16 +2 +8 -6
/LidCap3/ M16 +2 +9 -7
/LidCap4/ M16 +2 +6 -3
/LidCap5/ M16 +2 +7 -4
/DrainPort/ P10 +3
/VentPort/ P10 +4

VOLUMES UNITY
* Transport Cask Inner Lid Port Model - With Covers v1.5.1.0
PART 10 CLUSTER
ZROD M0 0.0000 0.0000 0.0000 1.2700 10.8966 ! Bottom cylinder
ZROD M0 0.0000 0.0000 10.8966 4.1275 7.5184 ! Middle cylinder
ZROD M16 0.0000 0.0000 18.4150 8.2931 2.5400 ! Cover
ZROD M0 0.0000 0.0000 20.9550 8.2931 1.9050 ! Top cylinder
ZROD M16 0.0000 0.0000 0.0000 8.2931 22.8600 ! Inner lid material

*Transport Cask - Normal Conditions v1.5.1.0
PART 11
ZROD 1 0.0000 0.0000 -34.6710 110.1090 490.1184 ! Transport Cask
ZROD 2 0.0000 0.0000 0.0000 90.1700 419.1000 ! Cavity
ZROD 3 0.0000 0.0000 -20.8280 100.1776 5.0800 ! Bottom neutron shield
ZROD 4 0.0000 0.0000 419.1000 100.3300 22.8600 ! Inner lid
ZROD 5 0.0000 0.0000 0.0000 103.4034 408.9400 ! Lead shield cavity
ZROD 6 0.0000 0.0000 0.0000 95.2500 30.4800 ! Inner shell lower
ZCON 7 0.0000 0.0000 30.4800 95.2500 93.9800 7.6200 ! Inner shell lower cone
ZROD 8 0.0000 0.0000 38.1000 93.9800 332.7400 ! Inner shell middle
ZCON 9 0.0000 0.0000 370.8400 93.9800 95.2500 7.6200 ! Inner shell upper cone
ZROD 10 0.0000 0.0000 378.4600 95.2500 30.4800 ! Inner shell upper
ZROD 11 0.0000 0.0000 0.0000 103.2891 408.9400 ! Lead shield
ZROD 12 0.0000 0.0000 -3.8100 124.7140 410.5148 ! Radial neutron shield shell
ZROD 13 0.0000 0.0000 -2.6100 124.0790 408.1148 ! Radial neutron shield
ZP 14 2.4700 ! Insulation (void) cut plane
ZCON 15 0.0000 0.0000 19.0500 124.7140 113.9190 24.1300 ! Top of rotating trunnion
BOX 16 -124.7140 -12.7000 -3.8100 249.4280 25.4000 22.8600 ! Bottom of rotating trunnion
BOX 17 -114.2238 -12.7000 -3.8100 228.4476 25.4000 22.8600 ! Bottom of rotating trunnion base
BOX 18 -124.7140 -7.6200 -3.8100 249.4280 15.2400 15.2400 ! Trunnion void box
XROD 19 -124.7140 0.0000 11.4300 7.6200 249.4280 ! Trunnion void circle
BOX 20 -124.7140 -12.7000 -3.8100 249.4280 25.4000 46.9900 ! Trunnion extent box
ZSEC 21 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 14.6584 15.3416 ! Heat fin 1
ZSEC 22 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 29.6584 30.3416 ! Heat fin 2
ZSEC 23 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 44.6584 45.3416 ! Heat fin 3
ZSEC 24 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 59.6584 60.3416 ! Heat fin 4

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

ZSEC	25	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	74.6584	75.3416	! Heat fin 5
ZSEC	26	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	89.6584	90.3416	! Heat fin 6
ZSEC	27	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	104.6584	105.3416	! Heat fin 7
ZSEC	28	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	119.6584	120.3416	! Heat fin 8
ZSEC	29	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	134.6584	135.3416	! Heat fin 9
ZSEC	30	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	149.6584	150.3416	! Heat fin 10
ZSEC	31	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	164.6584	165.3416	! Heat fin 11
ZSEC	32	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	179.6584	180.3416	! Heat fin 12
ZSEC	33	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	194.6584	195.3416	! Heat fin 13
ZSEC	34	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	209.6584	210.3416	! Heat fin 14
ZSEC	35	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	224.6584	225.3416	! Heat fin 15
ZSEC	36	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	239.6584	240.3416	! Heat fin 16
ZSEC	37	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	254.6584	255.3416	! Heat fin 17
ZSEC	38	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	269.6584	270.3416	! Heat fin 18
ZSEC	39	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	284.6584	285.3416	! Heat fin 19
ZSEC	40	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	299.6584	300.3416	! Heat fin 20
ZSEC	41	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	314.6584	315.3416	! Heat fin 21
ZSEC	42	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	329.6584	330.3416	! Heat fin 22
ZSEC	43	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	344.6584	345.3416	! Heat fin 23
ZSEC	44	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	359.6584	360.3416	! Heat fin 24
ZROD	45	0.0000	0.0000	425.6024	162.5600	85.0900				! Upper impact limiter
ZROD	46	0.0000	0.0000	-89.9160	162.5600	85.0900				! Lower impact limiter
ZROD	47	0.0000	0.0000	426.2374	161.9250	83.8200				! Inside upper limiter shell
ZROD	48	0.0000	0.0000	-89.2810	161.9250	83.8200				! Inside lower limiter shell
ZROD	49	0.0000	0.0000	425.6024	110.7440	30.4800				! Upper end cap
ZROD	50	0.0000	0.0000	-35.3060	110.7440	30.4800				! Lower end cap
ZROD	51	0.0000	0.0000	-89.9160	162.5600	600.6084				! Container
ZROD	52	0.0000	0.0000	-3.8100	125.7140	27.3677				! Surface detector #1
ZROD	53	0.0000	0.0000	23.5577	125.7140	27.3677				! Surface detector #2
ZROD	54	0.0000	0.0000	50.9253	125.7140	27.3677				! Surface detector #3
ZROD	55	0.0000	0.0000	78.2930	125.7140	27.3677				! Surface detector #4
ZROD	56	0.0000	0.0000	105.6606	125.7140	27.3677				! Surface detector #5
ZROD	57	0.0000	0.0000	133.0283	125.7140	27.3677				! Surface detector #6
ZROD	58	0.0000	0.0000	160.3959	125.7140	27.3677				! Surface detector #7
ZROD	59	0.0000	0.0000	187.7636	125.7140	27.3677				! Surface detector #8
ZROD	60	0.0000	0.0000	215.1312	125.7140	27.3677				! Surface detector #9
ZROD	61	0.0000	0.0000	242.4989	125.7140	27.3677				! Surface detector #10
ZROD	62	0.0000	0.0000	269.8665	125.7140	27.3677				! Surface detector #11
ZROD	63	0.0000	0.0000	297.2342	125.7140	27.3677				! Surface detector #12
ZROD	64	0.0000	0.0000	324.6018	125.7140	27.3677				! Surface detector #13
ZROD	65	0.0000	0.0000	351.9695	125.7140	27.3677				! Surface detector #14
ZROD	66	0.0000	0.0000	379.3371	125.7140	27.3677				! Surface detector #15
ZROD	67	0.0000	0.0000	406.7048	111.1090	18.8976				! Package detector betw. NS and Uplim
ZSEC	68	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	2.5000	7.5000	! Heat fin azi detector #1
ZSEC	69	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	7.5000	12.5000	! Heat fin azi detector #2
ZSEC	70	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	12.5000	17.5000	! Heat fin azi detector #3
ZSEC	71	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	17.5000	22.5000	! Heat fin azi detector #4
ZSEC	72	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	22.5000	27.5000	! Heat fin azi detector #5
ZSEC	73	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	27.5000	32.5000	! Heat fin azi detector #6
ZSEC	74	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	32.5000	37.5000	! Heat fin azi detector #7
ZSEC	75	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	37.5000	42.5000	! Heat fin azi detector #8
ZSEC	76	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	42.5000	47.5000	! Heat fin azi detector #9
ZSEC	77	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	47.5000	52.5000	! Heat fin azi detector #10
ZSEC	78	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	52.5000	57.5000	! Heat fin azi detector #11
ZSEC	79	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	57.5000	62.5000	! Heat fin azi detector #12
ZSEC	80	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	62.5000	67.5000	! Heat fin azi detector #13
ZSEC	81	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	67.5000	72.5000	! Heat fin azi detector #14
ZSEC	82	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	72.5000	77.5000	! Heat fin azi detector #15
ZSEC	83	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	77.5000	82.5000	! Heat fin azi detector #16
ZSEC	84	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	82.5000	87.5000	! Heat fin azi detector #17
ZSEC	85	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	87.5000	92.5000	! Heat fin azi detector #18
ZSEC	86	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	92.5000	97.5000	! Heat fin azi detector #19
ZSEC	87	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	97.5000	102.5000	! Heat fin azi detector #20
ZSEC	88	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	102.5000	107.5000	! Heat fin azi detector #21
ZSEC	89	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	107.5000	112.5000	! Heat fin azi detector #22
ZSEC	90	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	112.5000	117.5000	! Heat fin azi detector #23
ZSEC	91	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	117.5000	122.5000	! Heat fin azi detector #24
ZSEC	92	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	122.5000	127.5000	! Heat fin azi detector #25
ZSEC	93	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	127.5000	132.5000	! Heat fin azi detector #26
ZSEC	94	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	132.5000	137.5000	! Heat fin azi detector #27
ZSEC	95	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	137.5000	142.5000	! Heat fin azi detector #28
ZSEC	96	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	142.5000	147.5000	! Heat fin azi detector #29
ZSEC	97	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	147.5000	152.5000	! Heat fin azi detector #30
ZSEC	98	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	152.5000	157.5000	! Heat fin azi detector #31

Energy Group 7 – Normal Conditions (continued)

5.5-8

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

ZSEC	173	0.0000	0.0000	-3.8100	126.7140	127.7140	22.8600	335.0000	345.0000	! Rotating trunnion det#34
ZSEC	174	0.0000	0.0000	-3.8100	126.7140	127.7140	22.8600	345.0000	355.0000	! Rotating trunnion det#35
ZSEC	175	0.0000	0.0000	-3.8100	126.7140	127.7140	22.8600	355.0000	365.0000	! Rotating trunnion det#36
ZROD	176	0.0000	0.0000	-4.8260	135.7630	430.4284				! Personnel barrier surface
ZROD	177	0.0000	0.0000	-4.8260	136.7630	28.6952				! Personnel barrier det #1
ZROD	178	0.0000	0.0000	23.8692	136.7630	28.6952				! Personnel barrier det #2
ZROD	179	0.0000	0.0000	52.5645	136.7630	28.6952				! Personnel barrier det #3
ZROD	180	0.0000	0.0000	81.2597	136.7630	28.6952				! Personnel barrier det #4
ZROD	181	0.0000	0.0000	109.9549	136.7630	28.6952				! Personnel barrier det #5
ZROD	182	0.0000	0.0000	138.6501	136.7630	28.6952				! Personnel barrier det #6
ZROD	183	0.0000	0.0000	167.3454	136.7630	28.6952				! Personnel barrier det #7
ZROD	184	0.0000	0.0000	196.0406	136.7630	28.6952				! Personnel barrier det #8
ZROD	185	0.0000	0.0000	224.7358	136.7630	28.6952				! Personnel barrier det #9
ZROD	186	0.0000	0.0000	253.4310	136.7630	28.6952				! Personnel barrier det #10
ZROD	187	0.0000	0.0000	282.1263	136.7630	28.6952				! Personnel barrier det #11
ZROD	188	0.0000	0.0000	310.8215	136.7630	28.6952				! Personnel barrier det #12
ZROD	189	0.0000	0.0000	339.5167	136.7630	28.6952				! Personnel barrier det #13
ZROD	190	0.0000	0.0000	368.2119	136.7630	28.6952				! Personnel barrier det #14
ZROD	191	0.0000	0.0000	396.9072	136.7630	28.6952				! Personnel barrier det #15
ZONES										
/Cavity/	P8	+2								
/OuterShell/	M16	+1	-2	-3	-4	-5				
/InnerShell1/	M16	+6	-2							
/InnerShell2/	M16	+7	-2							
/InnerShell3/	M16	+8	-2							
/InnerShell4/	M16	+9	-2							
/InnerShell5/	M16	+10	-2							
/InnerLid/	P9	+4								
/BotNSShield/	M15	+3								
/LeadShield/	M14	+11	-6	-7	-8	-9	-10			
/LeadShieldGap/	M0	+5	-11							
/RadNSShieldShell1/	M16	+12	-13	-1	-20					
/RadNSShieldShell2/	M16	+12	-13	-1	+20	-15	-16			
/RadNSShield1/	M15	+13	-1	-21	-22	-23	-24	-25		
		-26	-27	-28	-29	-30	-31	-32		
		-33	-34	-35	-36	-37	-38	-39		
		-40	-41	-42	-43	-44	+14	-20		
/RadNSShield2/	M15	+13	-1	-32	-44	+20	-15	-16		
/InsulationVoid/	M0	+13	-1	-14	-20					
/RotTrunUpper/	M16	+15	-1	+20	+12					
/RotTrunLower/	M16	+17	-1	+12						
/RotTrunSide/	M16	+16	+12	-17	-18	-19				
/RotTrunBoxVoid/	M0	+18	+12	-17						
/RotTrunCircVoid/	M0	+19	-18	-17	+12					
/HeatFin1/	M18	+21	+14							
/HeatFin2/	M18	+22	+14							
/HeatFin3/	M18	+23	+14							
/HeatFin4/	M18	+24	+14							
/HeatFin5/	M18	+25	+14							
/HeatFin6/	M18	+26	+14							
/HeatFin7/	M18	+27	+14							
/HeatFin8/	M18	+28	+14							
/HeatFin9/	M18	+29	+14							
/HeatFin10/	M18	+30	+14							
/HeatFin11/	M18	+31	+14							
/HeatFin12/	M18	+32	+14	-15	-16					
/HeatFin13/	M18	+33	+14							
/HeatFin14/	M18	+34	+14							
/HeatFin15/	M18	+35	+14							
/HeatFin16/	M18	+36	+14							
/HeatFin17/	M18	+37	+14							
/HeatFin18/	M18	+38	+14							
/HeatFin19/	M18	+39	+14							
/HeatFin20/	M18	+40	+14							
/HeatFin21/	M18	+41	+14							
/HeatFin22/	M18	+42	+14							
/HeatFin23/	M18	+43	+14							
/HeatFin24/	M18	+44	+14	-15	-16					
/UpLimShell/	M16	+45	-47	-49						
/UpLimEnd/	M16	+49	-1							
/LoLimShell/	M16	+46	-48	-50						
/LoLimEnd/	M16	+50	-1							
/UpBalsa/	M19	+47	-49							
/LoBalsa/	M19	+48	-50							

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

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/InsidePersBarr/  M0 +176 -1 -12 -45 -46 -67 -52
-53 -54 -55 -56 -57 -58 -59
-60 -61 -62 -63 -64 -65 -66
-68 -69 -70 -71 -72 -73 -74
-75 -76 -77 -78 -79 -80 -81
-82 -83 -84 -85 -86 -87 -88
-89 -90 -91 -92 -93 -94 -95
-96 -97 -98 -99 -100 -101 -102
-103 -104 -105 -106 -107 -108 -109
-110 -111 -112 -113 -114 -115 -116
-117 -118 -119 -120 -121 -122 -123
-124 -125 -126 -127 -128 -129 -130
-131 -132 -133 -134 -135 -136 -137
-138 -139 -140 -141 -142 -143 -144
-145 -146 -147 -148 -149 -150 -151
-152 -153 -154 -155 -156 -157 -158
-159 -160 -161 -162 -163 -164 -165
-166 -167 -168 -169 -170 -171 -172
-173 -174 -175

/SurfaceDet1/  M0 +52 -12
/SurfaceDet2/  M0 +53 -12
/SurfaceDet3/  M0 +54 -12
/SurfaceDet4/  M0 +55 -12
/SurfaceDet5/  M0 +56 -12
/SurfaceDet6/  M0 +57 -12
/SurfaceDet7/  M0 +58 -12
/SurfaceDet8/  M0 +59 -12
/SurfaceDet9/  M0 +60 -12
/SurfaceDet10/ M0 +61 -12
/SurfaceDet11/ M0 +62 -12
/SurfaceDet12/ M0 +63 -12
/SurfaceDet13/ M0 +64 -12
/SurfaceDet14/ M0 +65 -12
/SurfaceDet15/ M0 +66 -12
/PackageDet/   M0 +67 -1

/HeatFinAziDet1/ M0 +68
/HeatFinAziDet2/ M0 +69
/HeatFinAziDet3/ M0 +70
/HeatFinAziDet4/ M0 +71
/HeatFinAziDet5/ M0 +72
/HeatFinAziDet6/ M0 +73
/HeatFinAziDet7/ M0 +74
/HeatFinAziDet8/ M0 +75
/HeatFinAziDet9/ M0 +76
/HeatFinAziDet10/ M0 +77
/HeatFinAziDet11/ M0 +78
/HeatFinAziDet12/ M0 +79
/HeatFinAziDet13/ M0 +80
/HeatFinAziDet14/ M0 +81
/HeatFinAziDet15/ M0 +82
/HeatFinAziDet16/ M0 +83
/HeatFinAziDet17/ M0 +84
/HeatFinAziDet18/ M0 +85
/HeatFinAziDet19/ M0 +86
/HeatFinAziDet20/ M0 +87
/HeatFinAziDet21/ M0 +88
/HeatFinAziDet22/ M0 +89
/HeatFinAziDet23/ M0 +90
/HeatFinAziDet24/ M0 +91
/HeatFinAziDet25/ M0 +92
/HeatFinAziDet26/ M0 +93
/HeatFinAziDet27/ M0 +94
/HeatFinAziDet28/ M0 +95
/HeatFinAziDet29/ M0 +96
/HeatFinAziDet30/ M0 +97
/HeatFinAziDet31/ M0 +98
/HeatFinAziDet32/ M0 +99
/HeatFinAziDet33/ M0 +100
/HeatFinAziDet34/ M0 +101
/HeatFinAziDet35/ M0 +102
/HeatFinAziDet36/ M0 +103
/HeatFinAziDet37/ M0 +104
/HeatFinAziDet38/ M0 +105
/HeatFinAziDet39/ M0 +106

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Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

/HeatFinAziDet40/	M0	+107
/HeatFinAziDet41/	M0	+108
/HeatFinAziDet42/	M0	+109
/HeatFinAziDet43/	M0	+110
/HeatFinAziDet44/	M0	+111
/HeatFinAziDet45/	M0	+112
/HeatFinAziDet46/	M0	+113
/HeatFinAziDet47/	M0	+114
/HeatFinAziDet48/	M0	+115
/HeatFinAziDet49/	M0	+116
/HeatFinAziDet50/	M0	+117
/HeatFinAziDet51/	M0	+118
/HeatFinAziDet52/	M0	+119
/HeatFinAziDet53/	M0	+120
/HeatFinAziDet54/	M0	+121
/HeatFinAziDet55/	M0	+122
/HeatFinAziDet56/	M0	+123
/HeatFinAziDet57/	M0	+124
/HeatFinAziDet58/	M0	+125
/HeatFinAziDet59/	M0	+126
/HeatFinAziDet60/	M0	+127
/HeatFinAziDet61/	M0	+128
/HeatFinAziDet62/	M0	+129
/HeatFinAziDet63/	M0	+130
/HeatFinAziDet64/	M0	+131
/HeatFinAziDet65/	M0	+132
/HeatFinAziDet66/	M0	+133
/HeatFinAziDet67/	M0	+134
/HeatFinAziDet68/	M0	+135
/HeatFinAziDet69/	M0	+136
/HeatFinAziDet70/	M0	+137
/HeatFinAziDet71/	M0	+138
/HeatFinAziDet72/	M0	+139
/RotTrunDet1/	M0	+140
/RotTrunDet2/	M0	+141
/RotTrunDet3/	M0	+142
/RotTrunDet4/	M0	+143
/RotTrunDet5/	M0	+144
/RotTrunDet6/	M0	+145
/RotTrunDet7/	M0	+146
/RotTrunDet8/	M0	+147
/RotTrunDet9/	M0	+148
/RotTrunDet10/	M0	+149
/RotTrunDet11/	M0	+150
/RotTrunDet12/	M0	+151
/RotTrunDet13/	M0	+152
/RotTrunDet14/	M0	+153
/RotTrunDet15/	M0	+154
/RotTrunDet16/	M0	+155
/RotTrunDet17/	M0	+156
/RotTrunDet18/	M0	+157
/RotTrunDet19/	M0	+158
/RotTrunDet20/	M0	+159
/RotTrunDet21/	M0	+160
/RotTrunDet22/	M0	+161
/RotTrunDet23/	M0	+162
/RotTrunDet24/	M0	+163
/RotTrunDet25/	M0	+164
/RotTrunDet26/	M0	+165
/RotTrunDet27/	M0	+166
/RotTrunDet28/	M0	+167
/RotTrunDet29/	M0	+168
/RotTrunDet30/	M0	+169
/RotTrunDet31/	M0	+170
/RotTrunDet32/	M0	+171
/RotTrunDet33/	M0	+172
/RotTrunDet34/	M0	+173
/RotTrunDet35/	M0	+174
/RotTrunDet36/	M0	+175
/PersBarrDet1/	M0	+177 -176
/PersBarrDet2/	M0	+178 -176
/PersBarrDet3/	M0	+179 -176
/PersBarrDet4/	M0	+180 -176
/PersBarrDet5/	M0	+181 -176

Figure 5.5-2 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

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/PersBarrDet6/ M0 +182 -176
/PersBarrDet7/ M0 +183 -176
/PersBarrDet8/ M0 +184 -176
/PersBarrDet9/ M0 +185 -176
/PersBarrDet10/ M0 +186 -176
/PersBarrDet11/ M0 +187 -176
/PersBarrDet12/ M0 +188 -176
/PersBarrDet13/ M0 +189 -176
/PersBarrDet14/ M0 +190 -176
/PersBarrDet15/ M0 +191 -176
/Container/ M0 +51 -45 -46 -177 -178 -179 -180
-181 -182 -183 -184 -185 -186 -187
-188 -189 -190 -191
VOLUMES 52*1.0 15*2.1531E+04 1.3133E+04 72*3.9775E+03
36*5.0756E+02 15*2.4568E+04 1.0
* Transport Cask Detector Description v1.5.1.0
PART 12
* Radial Detector DRA (Surface) Bodies
* Radial Detector DRA+ (Package) Bodies
* Radial Detector DRAA (HeatFin) Bodies
* Radial Detector DRAB (RotTrun) Bodies
* Radial Detector DRB (PersBarr) Bodies
* Radial Detector DRC (LimSurf) Bodies
ZROD 1 0.0000 0.0000 -89.9160 162.5600 600.6084
ZROD 2 0.0000 0.0000 -89.9160 163.5600 40.0406
ZROD 3 0.0000 0.0000 -49.8754 163.5600 40.0406
ZROD 4 0.0000 0.0000 -9.8349 163.5600 40.0406
ZROD 5 0.0000 0.0000 30.2057 163.5600 40.0406
ZROD 6 0.0000 0.0000 70.2462 163.5600 40.0406
ZROD 7 0.0000 0.0000 110.2868 163.5600 40.0406
ZROD 8 0.0000 0.0000 150.3274 163.5600 40.0406
ZROD 9 0.0000 0.0000 190.3679 163.5600 40.0406
ZROD 10 0.0000 0.0000 230.4085 163.5600 40.0406
ZROD 11 0.0000 0.0000 270.4490 163.5600 40.0406
ZROD 12 0.0000 0.0000 310.4896 163.5600 40.0406
ZROD 13 0.0000 0.0000 350.5302 163.5600 40.0406
ZROD 14 0.0000 0.0000 390.5707 163.5600 40.0406
ZROD 15 0.0000 0.0000 430.6113 163.5600 40.0406
ZROD 16 0.0000 0.0000 470.6518 163.5600 40.0406
* Radial Detector DRD (1m) Bodies
ZROD 17 0.0000 0.0000 -189.9160 224.7140 800.6084
ZROD 18 0.0000 0.0000 -189.9160 225.7140 53.3739
ZROD 19 0.0000 0.0000 -136.5421 225.7140 53.3739
ZROD 20 0.0000 0.0000 -83.1682 225.7140 53.3739
ZROD 21 0.0000 0.0000 -29.7943 225.7140 53.3739
ZROD 22 0.0000 0.0000 23.5796 225.7140 53.3739
ZROD 23 0.0000 0.0000 76.9535 225.7140 53.3739
ZROD 24 0.0000 0.0000 130.3274 225.7140 53.3739
ZROD 25 0.0000 0.0000 183.7013 225.7140 53.3739
ZROD 26 0.0000 0.0000 237.0751 225.7140 53.3739
ZROD 27 0.0000 0.0000 290.4490 225.7140 53.3739
ZROD 28 0.0000 0.0000 343.8229 225.7140 53.3739
ZROD 29 0.0000 0.0000 397.1968 225.7140 53.3739
ZROD 30 0.0000 0.0000 450.5707 225.7140 53.3739
ZROD 31 0.0000 0.0000 503.9446 225.7140 53.3739
ZROD 32 0.0000 0.0000 557.3185 225.7140 53.3739
* Radial Detector DRE (1m-LimSurf) Bodies
ZROD 33 0.0000 0.0000 -239.9160 262.5600 900.6084
ZROD 34 0.0000 0.0000 -239.9160 263.5600 60.0406
ZROD 35 0.0000 0.0000 -179.8754 263.5600 60.0406
ZROD 36 0.0000 0.0000 -119.8349 263.5600 60.0406
ZROD 37 0.0000 0.0000 -59.7943 263.5600 60.0406
ZROD 38 0.0000 0.0000 0.2462 263.5600 60.0406
ZROD 39 0.0000 0.0000 60.2868 263.5600 60.0406
ZROD 40 0.0000 0.0000 120.3274 263.5600 60.0406
ZROD 41 0.0000 0.0000 180.3679 263.5600 60.0406
ZROD 42 0.0000 0.0000 240.4085 263.5600 60.0406
ZROD 43 0.0000 0.0000 300.4490 263.5600 60.0406
ZROD 44 0.0000 0.0000 360.4896 263.5600 60.0406
ZROD 45 0.0000 0.0000 420.5302 263.5600 60.0406
ZROD 46 0.0000 0.0000 480.5707 263.5600 60.0406
ZROD 47 0.0000 0.0000 540.6113 263.5600 60.0406
ZROD 48 0.0000 0.0000 600.6518 263.5600 60.0406
* Radial Detector DRF (2m+Railcar) Bodies

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

ZROD	49	0.0000	0.0000	-289.9160	357.4800	1000.6084
ZROD	50	0.0000	0.0000	-289.9160	358.4800	50.0304
ZROD	51	0.0000	0.0000	-239.8856	358.4800	50.0304
ZROD	52	0.0000	0.0000	-189.8552	358.4800	50.0304
ZROD	53	0.0000	0.0000	-139.8247	358.4800	50.0304
ZROD	54	0.0000	0.0000	-89.7943	358.4800	50.0304
ZROD	55	0.0000	0.0000	-39.7639	358.4800	50.0304
ZROD	56	0.0000	0.0000	10.2665	358.4800	50.0304
ZROD	57	0.0000	0.0000	60.2969	358.4800	50.0304
ZROD	58	0.0000	0.0000	110.3274	358.4800	50.0304
ZROD	59	0.0000	0.0000	160.3578	358.4800	50.0304
ZROD	60	0.0000	0.0000	210.3882	358.4800	50.0304
ZROD	61	0.0000	0.0000	260.4186	358.4800	50.0304
ZROD	62	0.0000	0.0000	310.4490	358.4800	50.0304
ZROD	63	0.0000	0.0000	360.4795	358.4800	50.0304
ZROD	64	0.0000	0.0000	410.5099	358.4800	50.0304
ZROD	65	0.0000	0.0000	460.5403	358.4800	50.0304
ZROD	66	0.0000	0.0000	510.5707	358.4800	50.0304
ZROD	67	0.0000	0.0000	560.6011	358.4800	50.0304
ZROD	68	0.0000	0.0000	610.6316	358.4800	50.0304
ZROD	69	0.0000	0.0000	660.6620	358.4800	50.0304
* Radial Detector DRG (4m) Bodies						
ZROD	70	0.0000	0.0000	-489.9160	524.7140	1400.6084
ZROD	71	0.0000	0.0000	-489.9160	525.7140	70.0304
ZROD	72	0.0000	0.0000	-419.8856	525.7140	70.0304
ZROD	73	0.0000	0.0000	-349.8552	525.7140	70.0304
ZROD	74	0.0000	0.0000	-279.8247	525.7140	70.0304
ZROD	75	0.0000	0.0000	-209.7943	525.7140	70.0304
ZROD	76	0.0000	0.0000	-139.7639	525.7140	70.0304
ZROD	77	0.0000	0.0000	-69.7335	525.7140	70.0304
ZROD	78	0.0000	0.0000	0.2969	525.7140	70.0304
ZROD	79	0.0000	0.0000	70.3274	525.7140	70.0304
ZROD	80	0.0000	0.0000	140.3578	525.7140	70.0304
ZROD	81	0.0000	0.0000	210.3882	525.7140	70.0304
ZROD	82	0.0000	0.0000	280.4186	525.7140	70.0304
ZROD	83	0.0000	0.0000	350.4490	525.7140	70.0304
ZROD	84	0.0000	0.0000	420.4795	525.7140	70.0304
ZROD	85	0.0000	0.0000	490.5099	525.7140	70.0304
ZROD	86	0.0000	0.0000	560.5403	525.7140	70.0304
ZROD	87	0.0000	0.0000	630.5707	525.7140	70.0304
ZROD	88	0.0000	0.0000	700.6011	525.7140	70.0304
ZROD	89	0.0000	0.0000	770.6316	525.7140	70.0304
ZROD	90	0.0000	0.0000	840.6620	525.7140	70.0304
* World						
ZROD	91	0.0000	0.0000	-539.9160	574.7140	1500.6084
* External Void						
ZROD	92	0.0000	0.0000	-589.9160	624.7140	1600.6084
ZONES						
/TransportCask/ P11 +1						
* Detector DRC (LimSurf)						
/DRC01/	M0	+2	-1			
/DRC02/	M0	+3	-1			
/DRC03/	M0	+4	-1			
/DRC04/	M0	+5	-1			
/DRC05/	M0	+6	-1			
/DRC06/	M0	+7	-1			
/DRC07/	M0	+8	-1			
/DRC08/	M0	+9	-1			
/DRC09/	M0	+10	-1			
/DRC10/	M0	+11	-1			
/DRC11/	M0	+12	-1			
/DRC12/	M0	+13	-1			
/DRC13/	M0	+14	-1			
/DRC14/	M0	+15	-1			
/DRC15/	M0	+16	-1			
/Void/	M0	+17	-1			
		-2	-3	-4	-5	-6
		-8	-9	-10	-11	-12
		-14	-15	-16		-13
* Detector DRD (1m)						
/DRD01/	M0	+18	-17			
/DRD02/	M0	+19	-17			
/DRD03/	M0	+20	-17			

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

/DRD04/	M0	+21	-17				
/DRD05/	M0	+22	-17				
/DRD06/	M0	+23	-17				
/DRD07/	M0	+24	-17				
/DRD08/	M0	+25	-17				
/DRD09/	M0	+26	-17				
/DRD10/	M0	+27	-17				
/DRD11/	M0	+28	-17				
/DRD12/	M0	+29	-17				
/DRD13/	M0	+30	-17				
/DRD14/	M0	+31	-17				
/DRD15/	M0	+32	-17				
/Void/	M0	+33	-17				
		-18	-19	-20	-21	-22	-23
		-24	-25	-26	-27	-28	-29
		-30	-31	-32			
* Detector DRE (1m+LimSurf)							
/DRE01/	M0	+34	-33				
/DRE02/	M0	+35	-33				
/DRE03/	M0	+36	-33				
/DRE04/	M0	+37	-33				
/DRE05/	M0	+38	-33				
/DRE06/	M0	+39	-33				
/DRE07/	M0	+40	-33				
/DRE08/	M0	+41	-33				
/DRE09/	M0	+42	-33				
/DRE10/	M0	+43	-33				
/DRE11/	M0	+44	-33				
/DRE12/	M0	+45	-33				
/DRE13/	M0	+46	-33				
/DRE14/	M0	+47	-33				
/DRE15/	M0	+48	-33				
/Void/	M0	+49	-33				
		-34	-35	-36	-37	-38	-39
		-40	-41	-42	-43	-44	-45
		-46	-47	-48			
* Detector DRF (2m+Railcar)							
/DRF01/	M0	+50	-49				
/DRF02/	M0	+51	-49				
/DRF03/	M0	+52	-49				
/DRF04/	M0	+53	-49				
/DRF05/	M0	+54	-49				
/DRF06/	M0	+55	-49				
/DRF07/	M0	+56	-49				
/DRF08/	M0	+57	-49				
/DRF09/	M0	+58	-49				
/DRF10/	M0	+59	-49				
/DRF11/	M0	+60	-49				
/DRF12/	M0	+61	-49				
/DRF13/	M0	+62	-49				
/DRF14/	M0	+63	-49				
/DRF15/	M0	+64	-49				
/DRF16/	M0	+65	-49				
/DRF17/	M0	+66	-49				
/DRF18/	M0	+67	-49				
/DRF19/	M0	+68	-49				
/DRF20/	M0	+69	-49				
/Void/	M0	+70	-49				
		-50	-51	-52	-53	-54	-55
		-56	-57	-58	-59	-60	-61
		-62	-63	-64	-65	-66	-67
		-68	-69				
* Detector DRG (4m)							
/DRG01/	M0	+71	-70				
/DRG02/	M0	+72	-70				
/DRG03/	M0	+73	-70				
/DRG04/	M0	+74	-70				
/DRG05/	M0	+75	-70				
/DRG06/	M0	+76	-70				
/DRG07/	M0	+77	-70				
/DRG08/	M0	+78	-70				
/DRG09/	M0	+79	-70				
/DRG10/	M0	+80	-70				
/DRG11/	M0	+81	-70				

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```

/DRG12/ M0      +82      -70
/DRG13/ M0      +83      -70
/DRG14/ M0      +84      -70
/DRG15/ M0      +85      -70
/DRG16/ M0      +86      -70
/DRG17/ M0      +87      -70
/DRG18/ M0      +88      -70
/DRG19/ M0      +89      -70
/DRG20/ M0      +90      -70
/Void/   M0      +91      -70
          -71      -72      -73      -74      -75      -76
          -77      -78      -79      -80      -81      -82
          -83      -84      -85      -86      -87      -88
          -89      -90
/ExtVoid/ M-2000 +92      -91
Volumes
          1.0      15*4.1023E+04      1.0      15*7.5527E+04      1.0      15*9.9238E+04
          1.0      20*1.1253E+05      1.0      20*2.3110E+05      1.0      1.0
end

*
* Unit 5 Splitting Geometry for Radial Detectors - Gamma
*
begin splitting geometry
  r   35   fill   0.0000
          n   10   89.9922
          n   1    93.9800
          n   16   103.4034
          n   1   110.1090
          n   5   124.0790
          n   1   124.7140
          n   1   167.5600

  z   48   fill   -94.9160
          n   1   -89.9160
          n   1   -89.2810
          n   2   -35.3060
          n   1   -34.6710
          n   2   -20.8280
          n   2   -15.7480
          n   3    0.0000
          n   5   18.6004
          n   13   387.4084
          n   2   410.2100
          n   3   419.1000
          !   n   0   419.1000
          n   3   434.3400
          n   2   439.4200
          n   3   455.4474
          n   1   456.0824
          n   2   510.0574
          n   1   510.6924
          n   1   515.6924

end

*
* Unit 6 - Source Geometry for Fuel Gamma
*
begin source geometry
  r   10   fill   0.0000   n   10   89.9922
  z   13
  18.6004   27.8206   37.0408   55.4812   92.3620   166.1236
  239.8852   276.7660   313.6468   350.5276   359.7478   368.9680
  378.1882   387.4084

end

*
* Unit 7
*
begin energy data
  gamma dice
  importance standard 22 groups

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```

scoring      as      importance
simple      source      histogram      weighting      automatic
end

*
* Unit 8 Importance Map - Radial
*
begin importance map
  calculate
  targets      15
  part      12
  zones
    2      3      4      5      6
    7      8      9      10      11
    12      13      14      15      16
  strengths
    1.0E+02      1.0E+02      5.0E+01      5.0E+01      5.0E+01
    1.0E+01      1.0E+01      1.0E+00      1.0E+01      1.0E+01
    5.0E+01      5.0E+01      5.0E+01      1.0E+02      1.0E+02
  defer mixing
  void density      0.10
  track
  !      coupled source
  !      write gamma importances to 32
  !      write unformatted file to 31
  !      use method d
end

*
* Unit 9 Scoring Data - Radial
*
begin scoring data
  flux
  part      11
  from      53      to      67      ! DRA
  some      68      ! DRA+
  from      69      to      140      ! DRAA
  from      141      to      176      ! DRAB
  from      177      to      191      ! DRB
  part      12
  from      2      to      16      ! DRC
  from      18      to      32      ! DRD
  from      34      to      48      ! DRE
  from      50      to      69      ! DRF
  from      71      to      90      ! DRG
  responses      sos      ditto
  contributions to responses      ditto
  ! score distribution for response
  ! weight distribution      total
end

*
* Unit 10 Response Data
*
begin response data
  * Scaled to mrem/hr
  /ansi ans-6.1.1-1977 photon flux-dose conversion factors - mcnp table h.2 - mrem/
  function pairs
    1.5000E+01      1.3300E-02
    1.4787E+01      1.3142E-02
    1.4577E+01      1.2985E-02
    1.4370E+01      1.2831E-02
    1.4166E+01      1.2678E-02
    1.3964E+01      1.2528E-02
    1.3766E+01      1.2379E-02
    1.3570E+01      1.2231E-02
    1.3377E+01      1.2086E-02
    1.3187E+01      1.1942E-02
    1.3000E+01      1.1800E-02
    1.2785E+01      1.1641E-02
    1.2573E+01      1.1483E-02
    1.2365E+01      1.1328E-02
    1.2160E+01      1.1175E-02
    1.1958E+01      1.1025E-02

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

1.1760E+01	1.0876E-02
1.1565E+01	1.0729E-02
1.1374E+01	1.0584E-02
1.1185E+01	1.0441E-02
1.1000E+01	1.0300E-02
1.0781E+01	1.0136E-02
1.0567E+01	9.9740E-03
1.0357E+01	9.8149E-03
1.0152E+01	9.6583E-03
9.9499E+00	9.5043E-03
9.7522E+00	9.3526E-03
9.5585E+00	9.2035E-03
9.3686E+00	9.0566E-03
9.1824E+00	8.9122E-03
9.0000E+00	8.7700E-03
8.8374E+00	8.6521E-03
8.6777E+00	8.5358E-03
8.5210E+00	8.4211E-03
8.3670E+00	8.3079E-03
8.2158E+00	8.1962E-03
8.0674E+00	8.0861E-03
7.9216E+00	7.9774E-03
7.7785E+00	7.8701E-03
7.6380E+00	7.7644E-03
7.5000E+00	7.6600E-03
7.4214E+00	7.6031E-03
7.3436E+00	7.5467E-03
7.2666E+00	7.4907E-03
7.1905E+00	7.4351E-03
7.1151E+00	7.3799E-03
7.0406E+00	7.3251E-03
6.9668E+00	7.2707E-03
6.8937E+00	7.2167E-03
6.8215E+00	7.1632E-03
6.7500E+00	7.1100E-03
6.6983E+00	7.0721E-03
6.6469E+00	7.0344E-03
6.5959E+00	6.9969E-03
6.5454E+00	6.9596E-03
6.4952E+00	6.9225E-03
6.4454E+00	6.8856E-03
6.3960E+00	6.8489E-03
6.3469E+00	6.8124E-03
6.2983E+00	6.7761E-03
6.2500E+00	6.7400E-03
6.1981E+00	6.7021E-03
6.1466E+00	6.6643E-03
6.0956E+00	6.6268E-03
6.0450E+00	6.5895E-03
5.9948E+00	6.5524E-03
5.9450E+00	6.5155E-03
5.8956E+00	6.4788E-03
5.8467E+00	6.4423E-03
5.7981E+00	6.4061E-03
5.7500E+00	6.3700E-03
5.6979E+00	6.3331E-03
5.6463E+00	6.2963E-03
5.5952E+00	6.2598E-03
5.5445E+00	6.2235E-03
5.4943E+00	6.1874E-03
5.4446E+00	6.1515E-03
5.3953E+00	6.1158E-03
5.3464E+00	6.0803E-03
5.2980E+00	6.0451E-03
5.2500E+00	6.0100E-03
5.2244E+00	5.9887E-03
5.1990E+00	5.9674E-03
5.1737E+00	5.9462E-03
5.1485E+00	5.9251E-03
5.1235E+00	5.9041E-03
5.0985E+00	5.8831E-03
5.0737E+00	5.8622E-03
5.0490E+00	5.8414E-03
5.0245E+00	5.8207E-03

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

5.0000E+00	5.8000E-03
4.9744E+00	5.7797E-03
4.9490E+00	5.7594E-03
4.9236E+00	5.7393E-03
4.8985E+00	5.7192E-03
4.8734E+00	5.6991E-03
4.8485E+00	5.6792E-03
4.8237E+00	5.6593E-03
4.7990E+00	5.6394E-03
4.7744E+00	5.6197E-03
4.7500E+00	5.6000E-03
4.6975E+00	5.5619E-03
4.6455E+00	5.5240E-03
4.5941E+00	5.4863E-03
4.5433E+00	5.4490E-03
4.4931E+00	5.4118E-03
4.4434E+00	5.3750E-03
4.3942E+00	5.3384E-03
4.3456E+00	5.3020E-03
4.2975E+00	5.2659E-03
4.2500E+00	5.2300E-03
4.1971E+00	5.1886E-03
4.1449E+00	5.1474E-03
4.0934E+00	5.1066E-03
4.0425E+00	5.0662E-03
3.9922E+00	5.0260E-03
3.9425E+00	4.9862E-03
3.8935E+00	4.9467E-03
3.8451E+00	4.9075E-03
3.7972E+00	4.8686E-03
3.7500E+00	4.8300E-03
3.6967E+00	4.7863E-03
3.6442E+00	4.7429E-03
3.5924E+00	4.7000E-03
3.5414E+00	4.6574E-03
3.4911E+00	4.6152E-03
3.4415E+00	4.5734E-03
3.3926E+00	4.5320E-03
3.3444E+00	4.4910E-03
3.2968E+00	4.4503E-03
3.2500E+00	4.4100E-03
3.2019E+00	4.3683E-03
3.1546E+00	4.3269E-03
3.1079E+00	4.2860E-03
3.0619E+00	4.2454E-03
3.0166E+00	4.2052E-03
2.9720E+00	4.1655E-03
2.9280E+00	4.1260E-03
2.8847E+00	4.0870E-03
2.8420E+00	4.0483E-03
2.8000E+00	4.0100E-03
2.7793E+00	3.9906E-03
2.7588E+00	3.9713E-03
2.7384E+00	3.9520E-03
2.7182E+00	3.9329E-03
2.6981E+00	3.9138E-03
2.6782E+00	3.8949E-03
2.6585E+00	3.8760E-03
2.6388E+00	3.8573E-03
2.6193E+00	3.8386E-03
2.6000E+00	3.8200E-03
2.5569E+00	3.7780E-03
2.5146E+00	3.7364E-03
2.4729E+00	3.6953E-03
2.4319E+00	3.6547E-03
2.3917E+00	3.6145E-03
2.3520E+00	3.5747E-03
2.3131E+00	3.5354E-03
2.2747E+00	3.4965E-03
2.2371E+00	3.4580E-03
2.2000E+00	3.4200E-03
2.1563E+00	3.3744E-03
2.1135E+00	3.3293E-03
2.0715E+00	3.2849E-03

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

2.0303E+00	3.2410E-03
1.9900E+00	3.1978E-03
1.9504E+00	3.1551E-03
1.9117E+00	3.1130E-03
1.8737E+00	3.0714E-03
1.8365E+00	3.0304E-03
1.8000E+00	2.9900E-03
1.7553E+00	2.9381E-03
1.7118E+00	2.8872E-03
1.6693E+00	2.8371E-03
1.6279E+00	2.7879E-03
1.5875E+00	2.7395E-03
1.5481E+00	2.6920E-03
1.5096E+00	2.6453E-03
1.4722E+00	2.5994E-03
1.4356E+00	2.5543E-03
1.4000E+00	2.5100E-03
1.3537E+00	2.4512E-03
1.3089E+00	2.3937E-03
1.2656E+00	2.3376E-03
1.2237E+00	2.2828E-03
1.1832E+00	2.2293E-03
1.1441E+00	2.1771E-03
1.1062E+00	2.1260E-03
1.0696E+00	2.0762E-03
1.0342E+00	2.0275E-03
1.0000E+00	1.9800E-03
9.7793E-01	1.9477E-03
9.5635E-01	1.9160E-03
9.3525E-01	1.8848E-03
9.1461E-01	1.8541E-03
8.9443E-01	1.8238E-03
8.7469E-01	1.7941E-03
8.5539E-01	1.7649E-03
8.3651E-01	1.7361E-03
8.1805E-01	1.7078E-03
8.0000E-01	1.6800E-03
7.8939E-01	1.6633E-03
7.7892E-01	1.6467E-03
7.6859E-01	1.6303E-03
7.5839E-01	1.6141E-03
7.4833E-01	1.5980E-03
7.3841E-01	1.5821E-03
7.2861E-01	1.5663E-03
7.1895E-01	1.5507E-03
7.0941E-01	1.5353E-03
7.0000E-01	1.5200E-03
6.9483E-01	1.5118E-03
6.8970E-01	1.5037E-03
6.8461E-01	1.4955E-03
6.7955E-01	1.4875E-03
6.7454E-01	1.4795E-03
6.6956E-01	1.4715E-03
6.6461E-01	1.4635E-03
6.5971E-01	1.4557E-03
6.5483E-01	1.4478E-03
6.5000E-01	1.4400E-03
6.4482E-01	1.4318E-03
6.3968E-01	1.4236E-03
6.3458E-01	1.4155E-03
6.2952E-01	1.4075E-03
6.2450E-01	1.3994E-03
6.1952E-01	1.3915E-03
6.1458E-01	1.3835E-03
6.0968E-01	1.3756E-03
6.0482E-01	1.3678E-03
6.0000E-01	1.3600E-03
5.9480E-01	1.3507E-03
5.8965E-01	1.3415E-03
5.8454E-01	1.3324E-03
5.7948E-01	1.3233E-03
5.7446E-01	1.3142E-03
5.6948E-01	1.3053E-03
5.6455E-01	1.2964E-03

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

5.5966E-01	1.2875E-03
5.5481E-01	1.2787E-03
5.5000E-01	1.2700E-03
5.4478E-01	1.2596E-03
5.3962E-01	1.2493E-03
5.3450E-01	1.2391E-03
5.2943E-01	1.2290E-03
5.2440E-01	1.2190E-03
5.1943E-01	1.2090E-03
5.1450E-01	1.1991E-03
5.0962E-01	1.1893E-03
5.0479E-01	1.1796E-03
5.0000E-01	1.1700E-03
4.9476E-01	1.1607E-03
4.8957E-01	1.1514E-03
4.8444E-01	1.1422E-03
4.7937E-01	1.1331E-03
4.7434E-01	1.1241E-03
4.6937E-01	1.1151E-03
4.6445E-01	1.1062E-03
4.5958E-01	1.0974E-03
4.5477E-01	1.0887E-03
4.5000E-01	1.0800E-03
4.4473E-01	1.0701E-03
4.3952E-01	1.0603E-03
4.3438E-01	1.0506E-03
4.2929E-01	1.0409E-03
4.2426E-01	1.0314E-03
4.1930E-01	1.0220E-03
4.1439E-01	1.0126E-03
4.0953E-01	1.0033E-03
4.0474E-01	9.9411E-04
4.0000E-01	9.8500E-04
3.9469E-01	9.7374E-04
3.8946E-01	9.6260E-04
3.8429E-01	9.5160E-04
3.7920E-01	9.4072E-04
3.7417E-01	9.2996E-04
3.6920E-01	9.1933E-04
3.6431E-01	9.0882E-04
3.5947E-01	8.9843E-04
3.5470E-01	8.8815E-04
3.5000E-01	8.7800E-04
3.4465E-01	8.6531E-04
3.3937E-01	8.5279E-04
3.3418E-01	8.4046E-04
3.2907E-01	8.2831E-04
3.2404E-01	8.1633E-04
3.1908E-01	8.0453E-04
3.1420E-01	7.9290E-04
3.0939E-01	7.8143E-04
3.0466E-01	7.7014E-04
3.0000E-01	7.5900E-04
2.9458E-01	7.4511E-04
2.8926E-01	7.3147E-04
2.8403E-01	7.1809E-04
2.7890E-01	7.0495E-04
2.7386E-01	6.9205E-04
2.6891E-01	6.7938E-04
2.6405E-01	6.6695E-04
2.5928E-01	6.5474E-04
2.5460E-01	6.4276E-04
2.5000E-01	6.3100E-04
2.4448E-01	6.1661E-04
2.3909E-01	6.0255E-04
2.3381E-01	5.8881E-04
2.2865E-01	5.7538E-04
2.2361E-01	5.6226E-04
2.1867E-01	5.4943E-04
2.1385E-01	5.3690E-04
2.0913E-01	5.2466E-04
2.0451E-01	5.1269E-04
2.0000E-01	5.0100E-04
1.9433E-01	4.8721E-04

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

1.8882E-01	4.7380E-04
1.8346E-01	4.6076E-04
1.7826E-01	4.4808E-04
1.7321E-01	4.3575E-04
1.6829E-01	4.2376E-04
1.6352E-01	4.1210E-04
1.5888E-01	4.0075E-04
1.5438E-01	3.8973E-04
1.5000E-01	3.7900E-04
1.4404E-01	3.6809E-04
1.3832E-01	3.5749E-04
1.3282E-01	3.4720E-04
1.2754E-01	3.3721E-04
1.2247E-01	3.2750E-04
1.1761E-01	3.1807E-04
1.1293E-01	3.0892E-04
1.0845E-01	3.0002E-04
1.0414E-01	2.9139E-04
1.0000E-01	2.8300E-04
9.6496E-02	2.8039E-04
9.3115E-02	2.7781E-04
8.9852E-02	2.7526E-04
8.6704E-02	2.7272E-04
8.3666E-02	2.7021E-04
8.0734E-02	2.6772E-04
7.7906E-02	2.6526E-04
7.5176E-02	2.6282E-04
7.2542E-02	2.6040E-04
7.0000E-02	2.5800E-04
6.7684E-02	2.6103E-04
6.5444E-02	2.6410E-04
6.3279E-02	2.6721E-04
6.1185E-02	2.7035E-04
5.9161E-02	2.7353E-04
5.7203E-02	2.7675E-04
5.5311E-02	2.8000E-04
5.3481E-02	2.8330E-04
5.1711E-02	2.8663E-04
5.0000E-02	2.9000E-04
4.7510E-02	3.1092E-04
4.5144E-02	3.3335E-04
4.2896E-02	3.5740E-04
4.0760E-02	3.8318E-04
3.8730E-02	4.1083E-04
3.6801E-02	4.4047E-04
3.4968E-02	4.7224E-04
3.3227E-02	5.0631E-04
3.1572E-02	5.4284E-04
3.0000E-02	5.8200E-04
2.6879E-02	7.0502E-04
2.4082E-02	8.5404E-04
2.1577E-02	1.0346E-03
1.9332E-02	1.2532E-03
1.7321E-02	1.5181E-03
1.5518E-02	1.8390E-03
1.3904E-02	2.2277E-03
1.2457E-02	2.6986E-03
1.1161E-02	3.2690E-03
1.0000E-02	3.9600E-03

end

*
* Unit 13 Hole Data
*
begin hole data
* STC Basket Hole Description v1.2
* Hole 1 General Basket Structure
PLATE
0 0 1
7
417.8300 0 ! Top of Basket
389.4976 -2 ! Top of Highest Support Disk
326.4316 -7 ! Resume support disk only
79.2376 -4 ! Start of support+heat disk region

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```
17.4396 -6 ! Bottom of Lowest Support Disk
0.0000 -3 ! Bottom of Basket
0.0000 0 ! Basket Offset
0

* Hole 2 Top Weldment Disk - no structure above the weldment disk
RZMESH
2 ! number of radial points
85.5472
89.9922
3 ! number of axial intervals
389.4976 ! Top of diskstack
400.5580 ! Bottom of weldment
403.0980 ! Top of weldment plate
417.8300 ! Void to top of basket
0 0 ! Material below weldment
10 10 ! Plate Material
0 10 ! Flange
0 ! Outside material

* Hole 3 Bottom Weldment Disk - no structure in the weldment disk support
RZMESH
1 ! number of radial points
89.9922
1 ! number of axial intervals
3.8100
6.3500 ! Coordinates inherited from PLATE Hole
10 ! Plate Material
0 ! Outside material

* Hole 4 Support disk and heat transfer disk stack
PLATE
origin 0 0 79.2376 ! Origin
0 0 1
4
cell 12.3597 ! Sets up a repeating lattice of cells
12.3597 0 ! flood matl
7.6086 0 ! water gap
6.0211 11 ! aluminium disk
1.2700 0 ! water gap
9 ! steel disk

* Hole 5 Flood material model
PLATE
0 0 1
1
417.8300 0 ! Above flooded region
0 ! Flooded region

* Hole 6 Support disk stack lower
PLATE
origin 0 0 17.4396 ! Origin
0 0 1
2
cell 12.3596 ! Sets up a repeating lattice of cells
12.3596 0 ! flood matl
1.2700 0 ! water gap
9 ! steel disk

* Hole 7 Support disk stack upper
PLATE
origin 0 0 326.4316 ! Origin
0 0 1
2
cell 12.3596 ! Sets up a repeating lattice of cells
12.3596 0 ! flood matl
1.2700 0 ! water gap
9 ! steel disk

end

*
* Unit 15 Source Strength - Fuel Gamma
*
```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```
* Class 1 - aa14b - STC Hybrid14 (Rev 0) - Fuel Gamma - Group 7 Reponse
begin source strength
  component      6.9739E-06      : 1/volume (1/1.4339E+05)
  component      r      10*1.0
  component      z
    4.2500E-01      6.7500E-01      9.0000E-01      1.1000E+00      1.2000E+00      1.1500E+00
    1.0750E+00      1.0000E+00      8.6500E-01      7.4000E-01      6.5000E-01      5.5000E-01
    4.5000E-01
  component      energy
    6*0.0
    1.0000E+00
    15*0.0

end

*
* Unit 16 Simple Source Weights
*
*begin source weights
*
*end

*
* Unit 31 Tabular Output
*
begin tabular output
  /Case trnNrmDryRadFg_aa14b_07g - Det DRA - Surface - Response/
  response      interim
  number      some      1
  region      from      121      to      135
  output to file      also
  /Case trnNrmDryRadFg_aa14b_07g - Det DRB - PersBarr - Response/
  response
  number      some      1
  region      from      245      to      259
  output to file      also
  /Case trnNrmDryRadFg_aa14b_07g - Det DRC - LimSurf - Response/
  response
  number      some      1
  region      from      262      to      276
  output to file      also
  /Case trnNrmDryRadFg_aa14b_07g - Det DRD - 1m - Response/
  response
  number      some      1
  region      from      278      to      292
  output to file      also
  /Case trnNrmDryRadFg_aa14b_07g - Det DRE - 1m+LimSurf - Response/
  response
  number      some      1
  region      from      294      to      308
  output to file      also
end

*
* Unit 32 Material Specification
*
begin material specification
  type      gamma
  normalise
  nmixtures      3
  weight      mixture      1
    u235      3.2615E-02
    u238      8.4888E-01
    o      1.1850E-01
  atoms      mixture      2
    h      6.6667E-01
    o      3.3333E-01
  atoms      mixture      3
    c      2.8571E-01
    h      4.7619E-01
    o      2.3810E-01
*
* Materials List - Dry Conditions - v1.2 - Class 1 - aa14b - STC Hybrid14 (Rev 0) Fuel
*
```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```

nmaterials 20
volume      ! Homogenized aal4b Fuel
material 1
  mixture 1 density 10.4120 prop 3.1489E-01 ! UO2 mixture at 3.7%
  void prop 1.6671E-02 ! Gap
  zircalloy density 6.5500 prop 9.7331E-02 ! Tube, clad
  void prop 5.7111E-01 ! Interstitial, inside tubes
volume      ! Fuel pin cladding
material 2
  zircalloy density 6.5500 prop 1.0000
volume      ! Water In Lattice and Tube
material 3
  mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume      ! Water In Fuel Rod Clad Gap
material 4
  mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume      ! Lower Nozzle Material
material 5
  stainless 3041 steel density 7.9200 prop 0.2948
  zircalloy density 6.5500 prop 0.0831
  void prop 0.6220
volume      ! Upper Nozzle Material
material 6
  stainless 3041 steel density 7.9200 prop 0.3613
  void prop 0.6387
volume      ! Upper Plenum Material
material 7
  stainless 3041 steel density 7.9200 prop 0.1147
  zircalloy density 6.5500 prop 0.0858
  void prop 0.7995
*
* Materials List - Common Materials - v1.2
*
volume      ! Tube wall and cover sheet
material 8
  stainless 3041 steel density 7.9200 prop 1.0000
volume      ! Structural Disk Material
material 9
  stainless 3041 steel density 7.9200 prop 1.0000
volume      ! Weldment Material
material 10
  stainless 3041 steel density 7.9200 prop 1.0000
volume      ! Heat Transfer Disk Material
material 11
  aluminium prop 1.0000
volume      ! Canister Material
material 12
  stainless 3041 steel density 7.9200 prop 1.0000
atoms      ! Transfer steel
material 13 density 0 ! (SCALE carbon steel)
  c prop 3.9250E-03
  fe prop 8.3498E-02
volume      ! Lead
material 14
  pb density 11.0400 prop 1.0000
atoms      ! NS-4-FR
material 15 density 0 ! 0 means atom/b-cm
  b10 prop 8.5500E-05
  b11 prop 3.4200E-04
  al prop 7.8000E-03
  h prop 5.8500E-02
  o prop 2.6100E-02
  c prop 2.2600E-02
  n prop 1.3900E-03
volume      ! Stainless Steel 304
material 16
  stainless 3041 steel density 7.9200 prop 1.0000
volume      ! Vent port middle cylinder
material 17
  stainless 3041 steel density 7.9200 prop 0.5000
  void prop 0.5000
volume      ! Heat fins for transport cask
material 18

```

Figure 5.5-2 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 7 – Normal Conditions (continued)

```
cu          density 8.9200    prop 0.4286
stainless 304l steel      density 7.9200    prop 0.5714
volume
material 19
mixture 3 density 0.1250    prop 1.0000
volume
material 20
mixture 3 density 0.3870    prop 1.0000
end
```

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions

```
columns 1 200
*
* NAC-STC - aa14b_02g - Fuel Neutron - Radial
* Dry Cavity Conditions
* Normal Transport Conditions
* Transport Model Revision v1.5.1.0
* Shielding Revision v1.5
* STC Source Profile
* Cobalt Concentration of 1.2 g/kg
* Fuel Assembly Shift = Cavity, Basket Shift = None
*
* Parameters
*
@samps = 5000000
*
* Unit 1 Control Data
*
begin control data
run
sample limit @samps
time limit 1000m
seeds 61444 14676
chime every [@samps/10] samples
report interim results
sbd 30s
dump intervals 1
end
*
* Unit 3 Output Control
*
*begin output control
* suppress inflows
*end
*
* Unit 4 Material Geometry
*
begin material geometry
* Fuel Assembly Type A - Class 1 - aa14b - STC Hybrid14 (Rev 0)
PART 1 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly Type B - Class 1 - aa14b - STC Hybrid14 (Rev 0)
PART 2 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly in Tube (Type A) v1.1
PART 3
BOX 1 1.4135 1.4135 9.9060 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of basket cavity
ZONES
/Fuel Assembly/ P1 +1
/Space in Tube/ H5 +2 -1
/Fuel Tube/ M8 +3 -2
/Container/ H5 +4 -3 -2 -1
VOLUMES UNITY
* Fuel Assembly in Tube (Type B) v1.1
PART 4
BOX 1 1.4135 1.4135 9.9060 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of basket cavity
```


Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```

ZONES
/Fuel Assembly/   P2   +1
/Space in Tube/   H5   +2   -1
/Fuel Tube/       M8   +3   -2
/Container/       H5   +4   -3   -2   -1
VOLUMES UNITY
* Type A Disk Opening with Tube v1.2
PART   5   CLUSTER
BOX   P3   0.4547   0.4547   0.0000   22.5450   22.5450   419.1000   ! Fuel tube type A with fuel assy
BOX   H5   0.0000   0.0000   0.0000   23.4544   23.4544   419.1000   ! Support disk opening width
* Type B Disk Opening with Tube v1.2
PART   6   CLUSTER
BOX   P4   0.4547   0.4547   0.0000   22.5450   22.5450   419.1000   ! Fuel tube type B with fuel assy
BOX   H5   0.0000   0.0000   0.0000   23.4544   23.4544   419.1000   ! Support disk opening width
* STC Basket v1.2
PART   7
BOX   1   -38.9153   56.2432   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 1
BOX   2   -11.7272   56.2432   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 2
BOX   3   15.4610   56.2432   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 3
BOX   4   -70.7136   29.0551   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 4
BOX   5   -38.9153   29.0551   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 5
BOX   6   -11.7272   29.0551   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 6
BOX   7   15.4610   29.0551   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 7
BOX   8   47.2592   29.0551   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 8
BOX   9   -70.7136   1.8669   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 9
BOX   10  -38.9153   1.8669   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 10
BOX   11  -11.7272   1.8669   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 11
BOX   12  15.4610   1.8669   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 12
BOX   13  47.2592   1.8669   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 13
BOX   14  -70.7136   -25.3213   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 14
BOX   15  -38.9153   -25.3213   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 15
BOX   16  -11.7272   -25.3213   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 16
BOX   17  15.4610   -25.3213   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 17
BOX   18  47.2592   -25.3213   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 18
BOX   19  -70.7136   -52.5094   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 19
BOX   20  -38.9153   -52.5094   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 20
BOX   21  -11.7272   -52.5094   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 21
BOX   22  15.4610   -52.5094   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 22
BOX   23  47.2592   -52.5094   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 23
BOX   24  -38.9153   -79.6976   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 24
BOX   25  -11.7272   -79.6976   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 25
BOX   26  15.4610   -79.6976   0.0000   23.4544   23.4544   419.1000   ! Basket Opening 26
ZROD   27  0.0000   0.0000   0.0000   89.9922   419.1000   ! Basket stack to cavity height
ZONES
/Opening01/   P5   +1
/Opening02/   P5   +2
/Opening03/   P5   +3
/Opening04/   P6   +4
/Opening05/   P5   +5
/Opening06/   P5   +6
/Opening07/   P5   +7
/Opening08/   P6   +8
/Opening09/   P5   +9
/Opening10/   P5   +10
/Opening11/   P5   +11
/Opening12/   P5   +12
/Opening13/   P5   +13
/Opening14/   P5   +14
/Opening15/   P5   +15
/Opening16/   P5   +16
/Opening17/   P5   +17
/Opening18/   P5   +18
/Opening19/   P6   +19
/Opening20/   P5   +20
/Opening21/   P5   +21
/Opening22/   P5   +22
/Opening23/   P6   +23
/Opening24/   P5   +24
/Opening25/   P5   +25
/Opening26/   P5   +26
/Basket/      H1   +27   -1   -2   -3   -4   -5
              -6   -7   -8   -9   -10  -11
              -12  -13  -14  -15  -16  -17

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Figure 5.5-3 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

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-18 -19 -20 -21 -22 -23
-24 -25 -26
VOLUMES UNITY
* Basket in Cask Cavity v1.2
PART 8 NEST
ZROD P7 0.0000 0.0000 0.0000 89.9922 419.1000 ! Basket inserted - Includes gap to lid
ZROD H5 0.0000 0.0000 0.0000 90.1700 419.1000 ! Inserts flood matl to id of stc
* Transport Cask Inner Lid - With Ports v1.5.1.0
PART 9
ZROD 1 0.0000 0.0000 0.0000 100.3300 18.0848 ! Inner lid base
ZROD 2 0.0000 0.0000 18.0848 92.5957 4.7752 ! Inner lid cap
ZROD 3 0.0000 77.9907 0.0000 8.2931 22.8600 ! Drain port
ZROD 4 0.0000 -77.9907 0.0000 8.2931 22.8600 ! Vent port
ZROD 5 0.0000 0.0000 15.2400 85.6234 5.0800 ! Neutron shield
ZROD 6 0.0000 77.9907 15.2400 10.1600 5.0800 ! Cut circle 1 for neutron shield
ZROD 7 0.0000 -77.9907 15.2400 10.1600 5.0800 ! Cut circle 2 for neutron shield
BOX 8 -10.1600 77.9907 15.2400 20.3200 7.6327 5.0800 ! Cut box 1 for neutron shield
BOX 9 -10.1600 -85.6234 15.2400 20.3200 7.6327 5.0800 ! Cut box 2 for neutron shield
ZROD 10 0.0000 0.0000 0.0000 100.3300 22.8600 ! Container
ZONES
/Container/ M0 +10 -1 -2
/LidBase1/ M16 +1 -3 -4 -5 -6 -7
-8 -9
/LidBase2/ M16 +1 +8 -6
/LidBase3/ M16 +1 +9 -7
/LidBase4/ M16 +1 +6 -3
/LidBase5/ M16 +1 +7 -4
/Nshield/ M15 +5 -6 -7 -8 -9
/LidCap1/ M16 +2 -3 -4 -5 -6 -7
-8 -9
/LidCap2/ M16 +2 +8 -6
/LidCap3/ M16 +2 +9 -7
/LidCap4/ M16 +2 +6 -3
/LidCap5/ M16 +2 +7 -4
/DrainPort/ P10 +3
/VentPort/ P10 +4
VOLUMES UNITY
* Transport Cask Inner Lid Port Model - With Covers v1.5.1.0
PART 10 CLUSTER
ZROD M0 0.0000 0.0000 0.0000 1.2700 10.8966 ! Bottom cylinder
ZROD M0 0.0000 0.0000 10.8966 4.1275 7.5184 ! Middle cylinder
ZROD M16 0.0000 0.0000 18.4150 8.2931 2.5400 ! Cover
ZROD M0 0.0000 0.0000 20.9550 8.2931 1.9050 ! Top cylinder
ZROD M16 0.0000 0.0000 0.0000 8.2931 22.8600 ! Inner lid material
*Transport Cask - Normal Conditions v1.5.1.0
PART 11
ZROD 1 0.0000 0.0000 -34.6710 110.1090 490.1184 ! Transport Cask
ZROD 2 0.0000 0.0000 0.0000 90.1700 419.1000 ! Cavity
ZROD 3 0.0000 0.0000 -20.8280 100.1776 5.0800 ! Bottom neutron shield
ZROD 4 0.0000 0.0000 419.1000 100.3300 22.8600 ! Inner lid
ZROD 5 0.0000 0.0000 0.0000 103.4034 408.9400 ! Lead shield cavity
ZROD 6 0.0000 0.0000 0.0000 95.2500 30.4800 ! Inner shell lower
ZCONE 7 0.0000 0.0000 30.4800 95.2500 93.9800 7.6200 ! Inner shell lower cone
ZROD 8 0.0000 0.0000 38.1000 93.9800 332.7400 ! Inner shell middle
ZCONE 9 0.0000 0.0000 370.8400 93.9800 95.2500 7.6200 ! Inner shell upper cone
ZROD 10 0.0000 0.0000 378.4600 95.2500 30.4800 ! Inner shell upper
ZROD 11 0.0000 0.0000 0.0000 103.2891 408.9400 ! Lead shield
ZROD 12 0.0000 0.0000 -3.8100 124.7140 410.5148 ! Radial neutron shield shell
ZROD 13 0.0000 0.0000 -2.6100 124.0790 408.1148 ! Radial neutron shield
ZP 14 2.4700 ! Insulation (void) cut plane
ZCONE 15 0.0000 0.0000 19.0500 124.7140 113.9190 24.1300 ! Top of rotating trunnion
BOX 16 -124.7140 -12.7000 -3.8100 249.4280 25.4000 22.8600 ! Bottom of rotating trunnion
BOX 17 -114.2238 -12.7000 -3.8100 228.4476 25.4000 22.8600 ! Bottom of rotating trunnion base
BOX 18 -124.7140 -7.6200 -3.8100 249.4280 15.2400 15.2400 ! Trunnion void box
XROD 19 -124.7140 0.0000 11.4300 7.6200 249.4280 ! Trunnion void circle
BOX 20 -124.7140 -12.7000 -3.8100 249.4280 25.4000 46.9900 ! Trunnion extent box
ZSEC 21 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 14.6584 15.3416 ! Heat fin 1
ZSEC 22 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 29.6584 30.3416 ! Heat fin 2
ZSEC 23 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 44.6584 45.3416 ! Heat fin 3
ZSEC 24 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 59.6584 60.3416 ! Heat fin 4
ZSEC 25 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 74.6584 75.3416 ! Heat fin 5
ZSEC 26 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 89.6584 90.3416 ! Heat fin 6
ZSEC 27 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 104.6584 105.3416 ! Heat fin 7
ZSEC 28 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 119.6584 120.3416 ! Heat fin 8

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Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

ZSEC	29	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	134.6584	135.3416	! Heat fin 9
ZSEC	30	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	149.6584	150.3416	! Heat fin 10
ZSEC	31	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	164.6584	165.3416	! Heat fin 11
ZSEC	32	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	179.6584	180.3416	! Heat fin 12
ZSEC	33	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	194.6584	195.3416	! Heat fin 13
ZSEC	34	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	209.6584	210.3416	! Heat fin 14
ZSEC	35	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	224.6584	225.3416	! Heat fin 15
ZSEC	36	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	239.6584	240.3416	! Heat fin 16
ZSEC	37	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	254.6584	255.3416	! Heat fin 17
ZSEC	38	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	269.6584	270.3416	! Heat fin 18
ZSEC	39	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	284.6584	285.3416	! Heat fin 19
ZSEC	40	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	299.6584	300.3416	! Heat fin 20
ZSEC	41	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	314.6584	315.3416	! Heat fin 21
ZSEC	42	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	329.6584	330.3416	! Heat fin 22
ZSEC	43	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	344.6584	345.3416	! Heat fin 23
ZSEC	44	0.0000	0.0000	-2.6100	110.1090	124.0790	408.1148	359.6584	360.3416	! Heat fin 24
ZROD	45	0.0000	0.0000	425.6024	162.5600	85.0900				! Upper impact limiter
ZROD	46	0.0000	0.0000	-89.9160	162.5600	85.0900				! Lower impact limiter
ZROD	47	0.0000	0.0000	426.2374	161.9250	83.8200				! Inside upper limiter shell
ZROD	48	0.0000	0.0000	-89.2810	161.9250	83.8200				! Inside lower limiter shell
ZROD	49	0.0000	0.0000	425.6024	110.7440	30.4800				! Upper end cap
ZROD	50	0.0000	0.0000	-35.3060	110.7440	30.4800				! Lower end cap
ZROD	51	0.0000	0.0000	-89.9160	162.5600	600.6084				! Container
ZROD	52	0.0000	0.0000	-3.8100	125.7140	27.3677				! Surface detector #1
ZROD	53	0.0000	0.0000	23.5577	125.7140	27.3677				! Surface detector #2
ZROD	54	0.0000	0.0000	50.9253	125.7140	27.3677				! Surface detector #3
ZROD	55	0.0000	0.0000	78.2930	125.7140	27.3677				! Surface detector #4
ZROD	56	0.0000	0.0000	105.6606	125.7140	27.3677				! Surface detector #5
ZROD	57	0.0000	0.0000	133.0283	125.7140	27.3677				! Surface detector #6
ZROD	58	0.0000	0.0000	160.3959	125.7140	27.3677				! Surface detector #7
ZROD	59	0.0000	0.0000	187.7636	125.7140	27.3677				! Surface detector #8
ZROD	60	0.0000	0.0000	215.1312	125.7140	27.3677				! Surface detector #9
ZROD	61	0.0000	0.0000	242.4989	125.7140	27.3677				! Surface detector #10
ZROD	62	0.0000	0.0000	269.8665	125.7140	27.3677				! Surface detector #11
ZROD	63	0.0000	0.0000	297.2342	125.7140	27.3677				! Surface detector #12
ZROD	64	0.0000	0.0000	324.6018	125.7140	27.3677				! Surface detector #13
ZROD	65	0.0000	0.0000	351.9695	125.7140	27.3677				! Surface detector #14
ZROD	66	0.0000	0.0000	379.3371	125.7140	27.3677				! Surface detector #15
ZROD	67	0.0000	0.0000	406.7048	111.1090	18.8976				! Package detector betw. NS and UpLim
ZSEC	68	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	2.5000	7.5000	! Heat fin azi detector #1
ZSEC	69	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	7.5000	12.5000	! Heat fin azi detector #2
ZSEC	70	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	12.5000	17.5000	! Heat fin azi detector #3
ZSEC	71	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	17.5000	22.5000	! Heat fin azi detector #4
ZSEC	72	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	22.5000	27.5000	! Heat fin azi detector #5
ZSEC	73	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	27.5000	32.5000	! Heat fin azi detector #6
ZSEC	74	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	32.5000	37.5000	! Heat fin azi detector #7
ZSEC	75	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	37.5000	42.5000	! Heat fin azi detector #8
ZSEC	76	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	42.5000	47.5000	! Heat fin azi detector #9
ZSEC	77	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	47.5000	52.5000	! Heat fin azi detector #10
ZSEC	78	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	52.5000	57.5000	! Heat fin azi detector #11
ZSEC	79	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	57.5000	62.5000	! Heat fin azi detector #12
ZSEC	80	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	62.5000	67.5000	! Heat fin azi detector #13
ZSEC	81	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	67.5000	72.5000	! Heat fin azi detector #14
ZSEC	82	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	72.5000	77.5000	! Heat fin azi detector #15
ZSEC	83	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	77.5000	82.5000	! Heat fin azi detector #16
ZSEC	84	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	82.5000	87.5000	! Heat fin azi detector #17
ZSEC	85	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	87.5000	92.5000	! Heat fin azi detector #18
ZSEC	86	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	92.5000	97.5000	! Heat fin azi detector #19
ZSEC	87	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	97.5000	102.5000	! Heat fin azi detector #20
ZSEC	88	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	102.5000	107.5000	! Heat fin azi detector #21
ZSEC	89	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	107.5000	112.5000	! Heat fin azi detector #22
ZSEC	90	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	112.5000	117.5000	! Heat fin azi detector #23
ZSEC	91	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	117.5000	122.5000	! Heat fin azi detector #24
ZSEC	92	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	122.5000	127.5000	! Heat fin azi detector #25
ZSEC	93	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	127.5000	132.5000	! Heat fin azi detector #26
ZSEC	94	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	132.5000	137.5000	! Heat fin azi detector #27
ZSEC	95	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	137.5000	142.5000	! Heat fin azi detector #28
ZSEC	96	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	142.5000	147.5000	! Heat fin azi detector #29
ZSEC	97	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	147.5000	152.5000	! Heat fin azi detector #30
ZSEC	98	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	152.5000	157.5000	! Heat fin azi detector #31
ZSEC	99	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	157.5000	162.5000	! Heat fin azi detector #32
ZSEC	100	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	162.5000	167.5000	! Heat fin azi detector #33
ZSEC	101	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	167.5000	172.5000	! Heat fin azi detector #34
ZSEC	102	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	172.5000	177.5000	! Heat fin azi detector #35

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from Energy Group 2 – Normal Conditions (continued)

ZSEC	103	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	177.5000	182.5000	!	Heat	fin	azi	detector	#36
ZSEC	104	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	182.5000	187.5000	!	Heat	fin	azi	detector	#37
ZSEC	105	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	187.5000	192.5000	!	Heat	fin	azi	detector	#38
ZSEC	106	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	192.5000	197.5000	!	Heat	fin	azi	detector	#39
ZSEC	107	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	197.5000	202.5000	!	Heat	fin	azi	detector	#40
ZSEC	108	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	202.5000	207.5000	!	Heat	fin	azi	detector	#41
ZSEC	109	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	207.5000	212.5000	!	Heat	fin	azi	detector	#42
ZSEC	110	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	212.5000	217.5000	!	Heat	fin	azi	detector	#43
ZSEC	111	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	217.5000	222.5000	!	Heat	fin	azi	detector	#44
ZSEC	112	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	222.5000	227.5000	!	Heat	fin	azi	detector	#45
ZSEC	113	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	227.5000	232.5000	!	Heat	fin	azi	detector	#46
ZSEC	114	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	232.5000	237.5000	!	Heat	fin	azi	detector	#47
ZSEC	115	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	237.5000	242.5000	!	Heat	fin	azi	detector	#48
ZSEC	116	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	242.5000	247.5000	!	Heat	fin	azi	detector	#49
ZSEC	117	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	247.5000	252.5000	!	Heat	fin	azi	detector	#50
ZSEC	118	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	252.5000	257.5000	!	Heat	fin	azi	detector	#51
ZSEC	119	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	257.5000	262.5000	!	Heat	fin	azi	detector	#52
ZSEC	120	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	262.5000	267.5000	!	Heat	fin	azi	detector	#53
ZSEC	121	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	267.5000	272.5000	!	Heat	fin	azi	detector	#54
ZSEC	122	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	272.5000	277.5000	!	Heat	fin	azi	detector	#55
ZSEC	123	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	277.5000	282.5000	!	Heat	fin	azi	detector	#56
ZSEC	124	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	282.5000	287.5000	!	Heat	fin	azi	detector	#57
ZSEC	125	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	287.5000	292.5000	!	Heat	fin	azi	detector	#58
ZSEC	126	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	292.5000	297.5000	!	Heat	fin	azi	detector	#59
ZSEC	127	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	297.5000	302.5000	!	Heat	fin	azi	detector	#60
ZSEC	128	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	302.5000	307.5000	!	Heat	fin	azi	detector	#61
ZSEC	129	0.0000	0.0000	44.3800	125.7140	126.7140	361.1248	307.5000	312.5000						

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

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ZROD 177 0.0000 0.0000 -4.8260 136.7630 28.6952 ! Personnel barrier det #1
ZROD 178 0.0000 0.0000 23.8692 136.7630 28.6952 ! Personnel barrier det #2
ZROD 179 0.0000 0.0000 52.5645 136.7630 28.6952 ! Personnel barrier det #3
ZROD 180 0.0000 0.0000 81.2597 136.7630 28.6952 ! Personnel barrier det #4
ZROD 181 0.0000 0.0000 109.9549 136.7630 28.6952 ! Personnel barrier det #5
ZROD 182 0.0000 0.0000 138.6501 136.7630 28.6952 ! Personnel barrier det #6
ZROD 183 0.0000 0.0000 167.3454 136.7630 28.6952 ! Personnel barrier det #7
ZROD 184 0.0000 0.0000 196.0406 136.7630 28.6952 ! Personnel barrier det #8
ZROD 185 0.0000 0.0000 224.7358 136.7630 28.6952 ! Personnel barrier det #9
ZROD 186 0.0000 0.0000 253.4310 136.7630 28.6952 ! Personnel barrier det #10
ZROD 187 0.0000 0.0000 282.1263 136.7630 28.6952 ! Personnel barrier det #11
ZROD 188 0.0000 0.0000 310.8215 136.7630 28.6952 ! Personnel barrier det #12
ZROD 189 0.0000 0.0000 339.5167 136.7630 28.6952 ! Personnel barrier det #13
ZROD 190 0.0000 0.0000 368.2119 136.7630 28.6952 ! Personnel barrier det #14
ZROD 191 0.0000 0.0000 396.9072 136.7630 28.6952 ! Personnel barrier det #15
ZONES
/Cavity/ P8 +2
/OuterShell/ M16 +1 -2 -3 -4 -5
/InnerShell1/ M16 +6 -2
/InnerShell2/ M16 +7 -2
/InnerShell3/ M16 +8 -2
/InnerShell4/ M16 +9 -2
/InnerShell5/ M16 +10 -2
/InnerLid/ P9 +4
/BotNShield/ M15 +3
/LeadShield/ M14 +11 -6 -7 -8 -9 -10
/LeadShieldGap/ M0 +5 -11
/RadNShieldShell1/ M16 +12 -13 -1 -20
/RadNShieldShell2/ M16 +12 -13 -1 +20 -15 -16
/RadNShield1/ M15 +13 -1 -21 -22 -23 -24 -25
-26 -27 -28 -29 -30 -31 -32
-33 -34 -35 -36 -37 -38 -39
-40 -41 -42 -43 -44 +14 -20
/RadNShield2/ M15 +13 -1 -32 -44 +20 -15 -16
/InsulationVoid/ M0 +13 -1 -14 -20
/RotTrunUpper/ M16 +15 -1 +20 +12
/RotTrunLower/ M16 +17 -1 +12
/RotTrunSide/ M16 +16 +12 -17 -18 -19
/RotTrunBoxVoid/ M0 +18 +12 -17
/RotTrunCircVoid/ M0 +19 -18 -17 +12
/HeatFin1/ M18 +21 +14
/HeatFin2/ M18 +22 +14
/HeatFin3/ M18 +23 +14
/HeatFin4/ M18 +24 +14
/HeatFin5/ M18 +25 +14
/HeatFin6/ M18 +26 +14
/HeatFin7/ M18 +27 +14
/HeatFin8/ M18 +28 +14
/HeatFin9/ M18 +29 +14
/HeatFin10/ M18 +30 +14
/HeatFin11/ M18 +31 +14
/HeatFin12/ M18 +32 +14 -15 -16
/HeatFin13/ M18 +33 +14
/HeatFin14/ M18 +34 +14
/HeatFin15/ M18 +35 +14
/HeatFin16/ M18 +36 +14
/HeatFin17/ M18 +37 +14
/HeatFin18/ M18 +38 +14
/HeatFin19/ M18 +39 +14
/HeatFin20/ M18 +40 +14
/HeatFin21/ M18 +41 +14
/HeatFin22/ M18 +42 +14
/HeatFin23/ M18 +43 +14
/HeatFin24/ M18 +44 +14 -15 -16
/UplimShell/ M16 +45 -47 -49
/UplimEnd/ M16 +49 -1
/LoLimShell/ M16 +46 -48 -50
/LoLimEnd/ M16 +50 -1
/UpBalsa/ M19 +47 -49
/LoBalsa/ M19 +48 -50
/InsidePersBarr/ M0 +176 -1 -12 -45 -46 -67 -52
-53 -54 -55 -56 -57 -58 -59
-60 -61 -62 -63 -64 -65 -66
-68 -69 -70 -71 -72 -73 -74

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Figure 5.5-3 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

-75	-76	-77	-78	-79	-80	-81
-82	-83	-84	-85	-86	-87	-88
-89	-90	-91	-92	-93	-94	-95
-96	-97	-98	-99	-100	-101	-102
-103	-104	-105	-106	-107	-108	-109
-110	-111	-112	-113	-114	-115	-116
-117	-118	-119	-120	-121	-122	-123
-124	-125	-126	-127	-128	-129	-130
-131	-132	-133	-134	-135	-136	-137
-138	-139	-140	-141	-142	-143	-144
-145	-146	-147	-148	-149	-150	-151
-152	-153	-154	-155	-156	-157	-158
-159	-160	-161	-162	-163	-164	-165
-166	-167	-168	-169	-170	-171	-172
-173	-174	-175				
/SurfaceDet1/	M0	+52	-12			
/SurfaceDet2/	M0	+53	-12			
/SurfaceDet3/	M0	+54	-12			
/SurfaceDet4/	M0	+55	-12			
/SurfaceDet5/	M0	+56	-12			
/SurfaceDet6/	M0	+57	-12			
/SurfaceDet7/	M0	+58	-12			
/SurfaceDet8/	M0	+59	-12			
/SurfaceDet9/	M0	+60	-12			
/SurfaceDet10/	M0	+61	-12			
/SurfaceDet11/	M0	+62	-12			
/SurfaceDet12/	M0	+63	-12			
/SurfaceDet13/	M0	+64	-12			
/SurfaceDet14/	M0	+65	-12			
/SurfaceDet15/	M0	+66	-12			
/PackageDet/	M0	+67	-1			
/HeatFinAziDet1/	M0	+68				
/HeatFinAziDet2/	M0	+69				
/HeatFinAziDet3/	M0	+70				
/HeatFinAziDet4/	M0	+71				
/HeatFinAziDet5/	M0	+72				
/HeatFinAziDet6/	M0	+73				
/HeatFinAziDet7/	M0	+74				
/HeatFinAziDet8/	M0	+75				
/HeatFinAziDet9/	M0	+76				
/HeatFinAziDet10/	M0	+77				
/HeatFinAziDet11/	M0	+78				
/HeatFinAziDet12/	M0	+79				
/HeatFinAziDet13/	M0	+80				
/HeatFinAziDet14/	M0	+81				
/HeatFinAziDet15/	M0	+82				
/HeatFinAziDet16/	M0	+83				
/HeatFinAziDet17/	M0	+84				
/HeatFinAziDet18/	M0	+85				
/HeatFinAziDet19/	M0	+86				
/HeatFinAziDet20/	M0	+87				
/HeatFinAziDet21/	M0	+88				
/HeatFinAziDet22/	M0	+89				
/HeatFinAziDet23/	M0	+90				
/HeatFinAziDet24/	M0	+91				
/HeatFinAziDet25/	M0	+92				
/HeatFinAziDet26/	M0	+93				
/HeatFinAziDet27/	M0	+94				
/HeatFinAziDet28/	M0	+95				
/HeatFinAziDet29/	M0	+96				
/HeatFinAziDet30/	M0	+97				
/HeatFinAziDet31/	M0	+98				
/HeatFinAziDet32/	M0	+99				
/HeatFinAziDet33/	M0	+100				
/HeatFinAziDet34/	M0	+101				
/HeatFinAziDet35/	M0	+102				
/HeatFinAziDet36/	M0	+103				
/HeatFinAziDet37/	M0	+104				
/HeatFinAziDet38/	M0	+105				
/HeatFinAziDet39/	M0	+106				
/HeatFinAziDet40/	M0	+107				
/HeatFinAziDet41/	M0	+108				
/HeatFinAziDet42/	M0	+109				
/HeatFinAziDet43/	M0	+110				

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

/HeatFinAziDet44/	M0	+111
/HeatFinAziDet45/	M0	+112
/HeatFinAziDet46/	M0	+113
/HeatFinAziDet47/	M0	+114
/HeatFinAziDet48/	M0	+115
/HeatFinAziDet49/	M0	+116
/HeatFinAziDet50/	M0	+117
/HeatFinAziDet51/	M0	+118
/HeatFinAziDet52/	M0	+119
/HeatFinAziDet53/	M0	+120
/HeatFinAziDet54/	M0	+121
/HeatFinAziDet55/	M0	+122
/HeatFinAziDet56/	M0	+123
/HeatFinAziDet57/	M0	+124
/HeatFinAziDet58/	M0	+125
/HeatFinAziDet59/	M0	+126
/HeatFinAziDet60/	M0	+127
/HeatFinAziDet61/	M0	+128
/HeatFinAziDet62/	M0	+129
/HeatFinAziDet63/	M0	+130
/HeatFinAziDet64/	M0	+131
/HeatFinAziDet65/	M0	+132
/HeatFinAziDet66/	M0	+133
/HeatFinAziDet67/	M0	+134
/HeatFinAziDet68/	M0	+135
/HeatFinAziDet69/	M0	+136
/HeatFinAziDet70/	M0	+137
/HeatFinAziDet71/	M0	+138
/HeatFinAziDet72/	M0	+139
/RotTrunDet1/	M0	+140
/RotTrunDet2/	M0	+141
/RotTrunDet3/	M0	+142
/RotTrunDet4/	M0	+143
/RotTrunDet5/	M0	+144
/RotTrunDet6/	M0	+145
/RotTrunDet7/	M0	+146
/RotTrunDet8/	M0	+147
/RotTrunDet9/	M0	+148
/RotTrunDet10/	M0	+149
/RotTrunDet11/	M0	+150
/RotTrunDet12/	M0	+151
/RotTrunDet13/	M0	+152
/RotTrunDet14/	M0	+153
/RotTrunDet15/	M0	+154
/RotTrunDet16/	M0	+155
/RotTrunDet17/	M0	+156
/RotTrunDet18/	M0	+157
/RotTrunDet19/	M0	+158
/RotTrunDet20/	M0	+159
/RotTrunDet21/	M0	+160
/RotTrunDet22/	M0	+161
/RotTrunDet23/	M0	+162
/RotTrunDet24/	M0	+163
/RotTrunDet25/	M0	+164
/RotTrunDet26/	M0	+165
/RotTrunDet27/	M0	+166
/RotTrunDet28/	M0	+167
/RotTrunDet29/	M0	+168
/RotTrunDet30/	M0	+169
/RotTrunDet31/	M0	+170
/RotTrunDet32/	M0	+171
/RotTrunDet33/	M0	+172
/RotTrunDet34/	M0	+173
/RotTrunDet35/	M0	+174
/RotTrunDet36/	M0	+175
/PersBarrDet1/	M0	+177 -176
/PersBarrDet2/	M0	+178 -176
/PersBarrDet3/	M0	+179 -176
/PersBarrDet4/	M0	+180 -176
/PersBarrDet5/	M0	+181 -176
/PersBarrDet6/	M0	+182 -176
/PersBarrDet7/	M0	+183 -176
/PersBarrDet8/	M0	+184 -176
/PersBarrDet9/	M0	+185 -176

Figure 5.5-3 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

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/PersBarrDet10/  M0  +186  -176
/PersBarrDet11/  M0  +187  -176
/PersBarrDet12/  M0  +188  -176
/PersBarrDet13/  M0  +189  -176
/PersBarrDet14/  M0  +190  -176
/PersBarrDet15/  M0  +191  -176
/Container/      M0  +51   -45   -46   -177  -178  -179  -180
                -181  -182  -183  -184  -185  -186  -187
                -188  -189  -190  -191
VOLUMES          52*1.0      15*2.1531E+04      1.3133E+04      72*3.9775E+03
                36*5.0756E+02      15*2.4568E+04      1.0
* Transport Cask Detector Description v1.5.1.0
PART            12
* Radial Detector DRA (Surface) Bodies
* Radial Detector DRA+ (Package) Bodies
* Radial Detector DRAB (HeatFin) Bodies
* Radial Detector DRAB (RotTrun) Bodies
* Radial Detector DRB (PersBarr) Bodies
* Radial Detector DRC (LimSurf) Bodies
ZROD            1            0.0000  0.0000  -89.9160  162.5600  600.6084
ZROD            2            0.0000  0.0000  -89.9160  163.5600  40.0406
ZROD            3            0.0000  0.0000  -49.8754  163.5600  40.0406
ZROD            4            0.0000  0.0000  -9.8349   163.5600  40.0406
ZROD            5            0.0000  0.0000  30.2057   163.5600  40.0406
ZROD            6            0.0000  0.0000  70.2462   163.5600  40.0406
ZROD            7            0.0000  0.0000  110.2868  163.5600  40.0406
ZROD            8            0.0000  0.0000  150.3274  163.5600  40.0406
ZROD            9            0.0000  0.0000  190.3679  163.5600  40.0406
ZROD           10            0.0000  0.0000  230.4085  163.5600  40.0406
ZROD           11            0.0000  0.0000  270.4490  163.5600  40.0406
ZROD           12            0.0000  0.0000  310.4896  163.5600  40.0406
ZROD           13            0.0000  0.0000  350.5302  163.5600  40.0406
ZROD           14            0.0000  0.0000  390.5707  163.5600  40.0406
ZROD           15            0.0000  0.0000  430.6113  163.5600  40.0406
ZROD           16            0.0000  0.0000  470.6518  163.5600  40.0406
* Radial Detector DRD (1m) Bodies
ZROD           17            0.0000  0.0000  -189.9160  224.7140  800.6084
ZROD           18            0.0000  0.0000  -189.9160  225.7140  53.3739
ZROD           19            0.0000  0.0000  -136.5421  225.7140  53.3739
ZROD           20            0.0000  0.0000  -83.1682  225.7140  53.3739
ZROD           21            0.0000  0.0000  -29.7943  225.7140  53.3739
ZROD           22            0.0000  0.0000  23.5796   225.7140  53.3739
ZROD           23            0.0000  0.0000  76.9535   225.7140  53.3739
ZROD           24            0.0000  0.0000  130.3274  225.7140  53.3739
ZROD           25            0.0000  0.0000  183.7013  225.7140  53.3739
ZROD           26            0.0000  0.0000  237.0751  225.7140  53.3739
ZROD           27            0.0000  0.0000  290.4490  225.7140  53.3739
ZROD           28            0.0000  0.0000  343.8229  225.7140  53.3739
ZROD           29            0.0000  0.0000  397.1968  225.7140  53.3739
ZROD           30            0.0000  0.0000  450.5707  225.7140  53.3739
ZROD           31            0.0000  0.0000  503.9446  225.7140  53.3739
ZROD           32            0.0000  0.0000  557.3185  225.7140  53.3739
* Radial Detector DRE (1m+LimSurf) Bodies
ZROD           33            0.0000  0.0000  -239.9160  262.5600  900.6084
ZROD           34            0.0000  0.0000  -239.9160  263.5600  60.0406
ZROD           35            0.0000  0.0000  -179.8754  263.5600  60.0406
ZROD           36            0.0000  0.0000  -119.8349  263.5600  60.0406
ZROD           37            0.0000  0.0000  -59.7943   263.5600  60.0406
ZROD           38            0.0000  0.0000  0.2462    263.5600  60.0406
ZROD           39            0.0000  0.0000  60.2868   263.5600  60.0406
ZROD           40            0.0000  0.0000  120.3274   263.5600  60.0406
ZROD           41            0.0000  0.0000  180.3679   263.5600  60.0406
ZROD           42            0.0000  0.0000  240.4085   263.5600  60.0406
ZROD           43            0.0000  0.0000  300.4490   263.5600  60.0406
ZROD           44            0.0000  0.0000  360.4896   263.5600  60.0406
ZROD           45            0.0000  0.0000  420.5302   263.5600  60.0406
ZROD           46            0.0000  0.0000  480.5707   263.5600  60.0406
ZROD           47            0.0000  0.0000  540.6113   263.5600  60.0406
ZROD           48            0.0000  0.0000  600.6518   263.5600  60.0406
* Radial Detector DRF (2m+Railcar) Bodies
ZROD           49            0.0000  0.0000  -289.9160  357.4800  1000.6084
ZROD           50            0.0000  0.0000  -289.9160  358.4800  50.0304
ZROD           51            0.0000  0.0000  -239.8856  358.4800  50.0304
ZROD           52            0.0000  0.0000  -189.8552  358.4800  50.0304

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Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

ZROD	53	0.0000	0.0000	-139.8247	358.4800	50.0304
ZROD	54	0.0000	0.0000	-89.7943	358.4800	50.0304
ZROD	55	0.0000	0.0000	-39.7639	358.4800	50.0304
ZROD	56	0.0000	0.0000	10.2665	358.4800	50.0304
ZROD	57	0.0000	0.0000	60.2969	358.4800	50.0304
ZROD	58	0.0000	0.0000	110.3274	358.4800	50.0304
ZROD	59	0.0000	0.0000	160.3578	358.4800	50.0304
ZROD	60	0.0000	0.0000	210.3882	358.4800	50.0304
ZROD	61	0.0000	0.0000	260.4186	358.4800	50.0304
ZROD	62	0.0000	0.0000	310.4490	358.4800	50.0304
ZROD	63	0.0000	0.0000	360.4795	358.4800	50.0304
ZROD	64	0.0000	0.0000	410.5099	358.4800	50.0304
ZROD	65	0.0000	0.0000	460.5403	358.4800	50.0304
ZROD	66	0.0000	0.0000	510.5707	358.4800	50.0304
ZROD	67	0.0000	0.0000	560.6011	358.4800	50.0304
ZROD	68	0.0000	0.0000	610.6316	358.4800	50.0304
ZROD	69	0.0000	0.0000	660.6620	358.4800	50.0304
* Radial Detector DRG (4m) Bodies						
ZROD	70	0.0000	0.0000	-489.9160	524.7140	1400.6084
ZROD	71	0.0000	0.0000	-489.9160	525.7140	70.0304
ZROD	72	0.0000	0.0000	-419.8856	525.7140	70.0304
ZROD	73	0.0000	0.0000	-349.8552	525.7140	70.0304
ZROD	74	0.0000	0.0000	-279.8247	525.7140	70.0304
ZROD	75	0.0000	0.0000	-209.7943	525.7140	70.0304
ZROD	76	0.0000	0.0000	-139.7639	525.7140	70.0304
ZROD	77	0.0000	0.0000	-69.7335	525.7140	70.0304
ZROD	78	0.0000	0.0000	0.2969	525.7140	70.0304
ZROD	79	0.0000	0.0000	70.3274	525.7140	70.0304
ZROD	80	0.0000	0.0000	140.3578	525.7140	70.0304
ZROD	81	0.0000	0.0000	210.3882	525.7140	70.0304
ZROD	82	0.0000	0.0000	280.4186	525.7140	70.0304
ZROD	83	0.0000	0.0000	350.4490	525.7140	70.0304
ZROD	84	0.0000	0.0000	420.4795	525.7140	70.0304
ZROD	85	0.0000	0.0000	490.5099	525.7140	70.0304
ZROD	86	0.0000	0.0000	560.5403	525.7140	70.0304
ZROD	87	0.0000	0.0000	630.5707	525.7140	70.0304
ZROD	88	0.0000	0.0000	700.6011	525.7140	70.0304
ZROD	89	0.0000	0.0000	770.6316	525.7140	70.0304
ZROD	90	0.0000	0.0000	840.6620	525.7140	70.0304
* World						
ZROD	91	0.0000	0.0000	-539.9160	574.7140	1500.6084
* External Void						
ZROD	92	0.0000	0.0000	-589.9160	624.7140	1600.6084
ZONES						
/TransportCask/ P11 +1						
* Detector DRC (LimSurf)						
/DRC01/	M0	+2	-1			
/DRC02/	M0	+3	-1			
/DRC03/	M0	+4	-1			
/DRC04/	M0	+5	-1			
/DRC05/	M0	+6	-1			
/DRC06/	M0	+7	-1			
/DRC07/	M0	+8	-1			
/DRC08/	M0	+9	-1			
/DRC09/	M0	+10	-1			
/DRC10/	M0	+11	-1			
/DRC11/	M0	+12	-1			
/DRC12/	M0	+13	-1			
/DRC13/	M0	+14	-1			
/DRC14/	M0	+15	-1			
/DRC15/	M0	+16	-1			
/Void/	M0	+17	-1			
		-2	-3	-4	-5	-6
		-8	-9	-10	-11	-12
		-14	-15	-16		-13
* Detector DRD (1m)						
/DRD01/	M0	+18	-17			
/DRD02/	M0	+19	-17			
/DRD03/	M0	+20	-17			
/DRD04/	M0	+21	-17			
/DRD05/	M0	+22	-17			
/DRD06/	M0	+23	-17			
/DRD07/	M0	+24	-17			

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

/DRD08/	M0	+25	-17				
/DRD09/	M0	+26	-17				
/DRD10/	M0	+27	-17				
/DRD11/	M0	+28	-17				
/DRD12/	M0	+29	-17				
/DRD13/	M0	+30	-17				
/DRD14/	M0	+31	-17				
/DRD15/	M0	+32	-17				
/Void/	M0	+33	-17				
		-18	-19	-20	-21	-22	-23
		-24	-25	-26	-27	-28	-29
		-30	-31	-32			
* Detector DRE (1m+LimSurf)							
/DRE01/	M0	+34	-33				
/DRE02/	M0	+35	-33				
/DRE03/	M0	+36	-33				
/DRE04/	M0	+37	-33				
/DRE05/	M0	+38	-33				
/DRE06/	M0	+39	-33				
/DRE07/	M0	+40	-33				
/DRE08/	M0	+41	-33				
/DRE09/	M0	+42	-33				
/DRE10/	M0	+43	-33				
/DRE11/	M0	+44	-33				
/DRE12/	M0	+45	-33				
/DRE13/	M0	+46	-33				
/DRE14/	M0	+47	-33				
/DRE15/	M0	+48	-33				
/Void/	M0	+49	-33				
		-34	-35	-36	-37	-38	-39
		-40	-41	-42	-43	-44	-45
		-46	-47	-48			
* Detector DRF (2m+Railcar)							
/DRF01/	M0	+50	-49				
/DRF02/	M0	+51	-49				
/DRF03/	M0	+52	-49				
/DRF04/	M0	+53	-49				
/DRF05/	M0	+54	-49				
/DRF06/	M0	+55	-49				
/DRF07/	M0	+56	-49				
/DRF08/	M0	+57	-49				
/DRF09/	M0	+58	-49				
/DRF10/	M0	+59	-49				
/DRF11/	M0	+60	-49				
/DRF12/	M0	+61	-49				
/DRF13/	M0	+62	-49				
/DRF14/	M0	+63	-49				
/DRF15/	M0	+64	-49				
/DRF16/	M0	+65	-49				
/DRF17/	M0	+66	-49				
/DRF18/	M0	+67	-49				
/DRF19/	M0	+68	-49				
/DRF20/	M0	+69	-49				
/Void/	M0	+70	-49				
		-50	-51	-52	-53	-54	-55
		-56	-57	-58	-59	-60	-61
		-62	-63	-64	-65	-66	-67
		-68	-69				
* Detector DRG (4m)							
/DRG01/	M0	+71	-70				
/DRG02/	M0	+72	-70				
/DRG03/	M0	+73	-70				
/DRG04/	M0	+74	-70				
/DRG05/	M0	+75	-70				
/DRG06/	M0	+76	-70				
/DRG07/	M0	+77	-70				
/DRG08/	M0	+78	-70				
/DRG09/	M0	+79	-70				
/DRG10/	M0	+80	-70				
/DRG11/	M0	+81	-70				
/DRG12/	M0	+82	-70				
/DRG13/	M0	+83	-70				
/DRG14/	M0	+84	-70				
/DRG15/	M0	+85	-70				

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

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/DRG16/  M0      +86      -70
/DRG17/  M0      +87      -70
/DRG18/  M0      +88      -70
/DRG19/  M0      +89      -70
/DRG20/  M0      +90      -70
/Void/    M0      +91      -70
          -71      -72      -73      -74      -75      -76
          -77      -78      -79      -80      -81      -82
          -83      -84      -85      -86      -87      -88
          -89      -90
/ExtVoid/ M-2000 +92      -91
Volumes
          1.0      15*4.1023E+04      1.0      15*7.5527E+04      1.0      15*9.9238E+04
          1.0      20*1.1253E+05      1.0      20*2.3110E+05      1.0      1.0
end

*
* Unit 5 Splitting Geometry for Radial Detectors - Neutron
*
begin splitting geometry
  r   29   fill   0.0000
        n   10   89.9922
        n    2   93.9800
        n    2  103.4034
        n    3  110.1090
        n   10  124.0790
        n    1  124.7140
        n    1  167.5600

  z   61   fill  -94.9160
        n    1  -89.9160
        n    1  -89.2810
        n    4  -35.3060
        n    1  -34.6710
        n    5  -20.8280
        n    4  -15.7480
        n    6   0.0000
        n    1  18.6004
        n   13  387.4084
        n    1  410.2100
        n    1  419.1000
        !   n    0  419.1000
        n    6  434.3400
        n    4  439.4200
        n    6  455.4474
        n    1  456.0824
        n    4  510.0574
        n    1  510.6924
        n    1  515.6924

end

*
* Unit 6 - Source Geometry for Fuel Neutron
*
begin source geometry
  r   10   fill   0.0000   n   10   89.9922
  z   13
    18.6004  27.8206  37.0408  55.4812  92.3620  166.1236
    239.8852  276.7660  313.6468  350.5276  359.7478  368.9680
    378.1882  387.4084

end

*
* Unit 7
*
begin energy data
  neutron
  thermal treatment none
  importance standard 28 groups
  scoring as importance
  simple source histogram weighting automatic
end

```

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```

*
* Unit 8 Importance Map - Radial
*
begin importance map
  calculate
  targets 15
  part 12
  zones
    2 3 4 5 6
    7 8 9 10 11
    12 13 14 15 16
  strengths
    1.0E+02 1.0E+02 5.0E+01 5.0E+01 5.0E+01
    1.0E+01 1.0E+01 1.0E+00 1.0E+01 1.0E+01
    5.0E+01 5.0E+01 5.0E+01 1.0E+02 1.0E+02
  defer mixing
  void density 0.10
  track
! coupled source
! write gamma importances to 32
! write unformatted file to 31
! use method d
end

*
* Unit 9 Scoring Data - Radial
*
begin scoring data
  flux
  part 11
  from 53 to 67 ! DRA
  some 68 ! DRA+
  from 69 to 140 ! DRAA
  from 141 to 176 ! DRAB
  from 177 to 191 ! DRB
  part 12
  from 2 to 16 ! DRC
  from 18 to 32 ! DRD
  from 34 to 48 ! DRE
  from 50 to 69 ! DRF
  from 71 to 90 ! DRG
  responses sos ditto
  contributions to responses ditto
! score distribution for response
! weight distribution total
end

*
* Unit 10 Response Data
*
begin response data
* Scale to mrem/hr
/nrcrp38 - ansi ans-6.1.1-1977 neutron flux-dose conversion factors - mcnp table h.1 - mrem/
function pairs
  2.0000E+01 2.2700E-01
  1.9299E+01 2.2502E-01
  1.8623E+01 2.2307E-01
  1.7970E+01 2.2112E-01
  1.7341E+01 2.1920E-01
  1.6733E+01 2.1729E-01
  1.6147E+01 2.1540E-01
  1.5581E+01 2.1353E-01
  1.5035E+01 2.1167E-01
  1.4508E+01 2.0983E-01
  1.4000E+01 2.0800E-01
  1.3537E+01 2.0090E-01
  1.3089E+01 1.9405E-01
  1.2656E+01 1.8743E-01
  1.2237E+01 1.8104E-01
  1.1832E+01 1.7486E-01
  1.1441E+01 1.6889E-01
  1.1062E+01 1.6313E-01
  1.0696E+01 1.5757E-01

```

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

1.0342E+01	1.5219E-01
1.0000E+01	1.4700E-01
7.0000E+00	1.4700E-01
6.7684E+00	1.4788E-01
6.5444E+00	1.4876E-01
6.3279E+00	1.4964E-01
6.1185E+00	1.5054E-01
5.9161E+00	1.5143E-01
5.7203E+00	1.5234E-01
5.5311E+00	1.5324E-01
5.3481E+00	1.5416E-01
5.1711E+00	1.5508E-01
5.0000E+00	1.5600E-01
4.6652E+00	1.5258E-01
4.3528E+00	1.4924E-01
4.0613E+00	1.4597E-01
3.7893E+00	1.4277E-01
3.5355E+00	1.3964E-01
3.2988E+00	1.3658E-01
3.0779E+00	1.3359E-01
2.8717E+00	1.3066E-01
2.6794E+00	1.2780E-01
2.5000E+00	1.2500E-01
2.2811E+00	1.2568E-01
2.0814E+00	1.2637E-01
1.8991E+00	1.2706E-01
1.7329E+00	1.2775E-01
1.5811E+00	1.2845E-01
1.4427E+00	1.2915E-01
1.3164E+00	1.2986E-01
1.2011E+00	1.3057E-01
1.0960E+00	1.3128E-01
1.0000E+00	1.3200E-01
9.3303E-01	1.2740E-01
8.7055E-01	1.2296E-01
8.1225E-01	1.1868E-01
7.5786E-01	1.1455E-01
7.0711E-01	1.1056E-01
6.5975E-01	1.0671E-01
6.1557E-01	1.0299E-01
5.7435E-01	9.9404E-02
5.3589E-01	9.5942E-02
5.0000E-01	9.2600E-02
4.2567E-01	8.0093E-02
3.6239E-01	6.9276E-02
3.0852E-01	5.9919E-02
2.6265E-01	5.1826E-02
2.2361E-01	4.4827E-02
1.9037E-01	3.8772E-02
1.6207E-01	3.3536E-02
1.3797E-01	2.9006E-02
1.1746E-01	2.5089E-02
1.0000E-01	2.1700E-02
7.9433E-02	1.8112E-02
6.3096E-02	1.5117E-02
5.0119E-02	1.2617E-02
3.9811E-02	1.0531E-02
3.1623E-02	8.7893E-03
2.5119E-02	7.3359E-03
1.9953E-02	6.1228E-03
1.5849E-02	5.1104E-03
1.2589E-02	4.2653E-03
1.0000E-02	3.5600E-03
7.9433E-03	3.5795E-03
6.3096E-03	3.5991E-03
5.0119E-03	3.6189E-03
3.9811E-03	3.6387E-03
3.1623E-03	3.6586E-03
2.5119E-03	3.6787E-03
1.9953E-03	3.6988E-03
1.5849E-03	3.7191E-03
1.2589E-03	3.7395E-03
1.0000E-03	3.7600E-03
7.9433E-04	3.8000E-03

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```

6.3096E-04      3.8405E-03
5.0119E-04      3.8814E-03
3.9811E-04      3.9227E-03
3.1623E-04      3.9644E-03
2.5119E-04      4.0066E-03
1.9953E-04      4.0493E-03
1.5849E-04      4.0924E-03
1.2589E-04      4.1360E-03
1.0000E-04      4.1800E-03
7.9433E-05      4.2147E-03
6.3096E-05      4.2496E-03
5.0119E-05      4.2849E-03
3.9811E-05      4.3204E-03
3.1623E-05      4.3563E-03
2.5119E-05      4.3924E-03
1.9953E-05      4.4289E-03
1.5849E-05      4.4656E-03
1.2589E-05      4.5026E-03
1.0000E-05      4.5400E-03
7.9433E-06      4.5319E-03
6.3096E-06      4.5239E-03
5.0119E-06      4.5159E-03
3.9811E-06      4.5078E-03
3.1623E-06      4.4998E-03
2.5119E-06      4.4918E-03
1.9953E-06      4.4839E-03
1.5849E-06      4.4759E-03
1.2589E-06      4.4679E-03
1.0000E-06      4.4600E-03
7.9433E-07      4.3739E-03
6.3096E-07      4.2894E-03
5.0119E-07      4.2066E-03
3.9811E-07      4.1254E-03
3.1623E-07      4.0458E-03
2.5119E-07      3.9677E-03
1.9953E-07      3.8910E-03
1.5849E-07      3.8159E-03
1.2589E-07      3.7423E-03
1.0000E-07      3.6700E-03
2.5000E-08      3.6700E-03

end

*
* Unit 13 Hole Data
*
begin hole data
* STC Basket Hole Description v1.2
* Hole 1 General Basket Structure
PLATE
0 0 1
7
417.8300 0 ! Top of Basket
389.4976 -2 ! Top of Highest Support Disk
326.4316 -7 ! Resume support disk only
79.2376 -4 ! Start of support+heat disk region
17.4396 -6 ! Bottom of Lowest Support Disk
0.0000 -3 ! Bottom of Basket
0.0000 0 ! Basket Offset
0

* Hole 2 Top Weldment Disk - no structure above the weldment disk
RZMESH
2 ! number of radial points
85.5472
89.9922
3 ! number of axial intervals
389.4976 ! Top of diskstack
400.5580 ! Bottom of weldment
403.0980 ! Top of weldment plate
417.8300 ! Void to top of basket
0 0 ! Material below weldment
10 10 ! Plate Material
0 10 ! Flange
0 ! Outside material

```

Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```
* Hole      3      Bottom Weldment Disk - no structure in the weldment disk support
RZMESH
1              ! number of radial points
89.9922
1              ! number of axial intervals
3.8100
6.3500          ! Coordinates inherited from PLATE Hole
10             ! Plate Material
0              ! Outside material

* Hole      4      Support disk and heat transfer disk stack
PLATE
origin 0 0 79.2376 ! Origin
0 0 1
4
cell 12.3597      ! Sets up a repeating lattice of cells
12.3597 0          ! flood matl
7.6086 0          ! water gap
6.0211 11         ! aluminium disk
1.2700 0          ! water gap
9                ! steel disk

* Hole      5      Flood material model
PLATE
0 0 1
1
417.8300 0        ! Above flooded region
0                ! Flooded region

* Hole      6      Support disk stack lower
PLATE
origin 0 0 17.4396 ! Origin
0 0 1
2
cell 12.3596      ! Sets up a repeating lattice of cells
12.3596 0          ! flood matl
1.2700 0          ! water gap
9                ! steel disk

* Hole      7      Support disk stack upper
PLATE
origin 0 0 326.4316 ! Origin
0 0 1
2
cell 12.3596      ! Sets up a repeating lattice of cells
12.3596 0          ! flood matl
1.2700 0          ! water gap
9                ! steel disk

end

*
* Unit 15 Source Strength - Fuel Neutron
*
* Class 1 - aa14b - STC Hybrid14 (Rev 0) - Fuel Neutron - Group 2 Reponse
begin source strength
component      1.6667E+00 ! Subcritical multiplication factor
component      6.9739E-06 ! 1/volume (1/1.4339E+05)
component      r      10*1.0
component      z
4.3222E-02    2.3510E-01    6.9499E-01    1.5792E+00    2.1585E+00    1.8268E+00
1.3619E+00    1.0170E+00    5.7791E-01    2.8622E-01    1.6890E-01    8.4742E-02
3.7293E-02
component      energy
1*0.0
1.0000E+00
26*0.0

end

*
* Unit 16 Simple Source Weights
```

Figure 5.5-3 MCBEND Input File for Directly Loaded 14x14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```

*
*begin source weights
*
*end

*
* Unit 31 Tabular Output
*
begin tabular output
/Case trnNrmDryRadFn_aal4b_02g - Det DRA - Surface - Response/
response interim
number some 1
region from 121 to 135
output to file also
/Case trnNrmDryRadFn_aal4b_02g - Det DRB - PersBarr - Response/
response
number some 1
region from 245 to 259
output to file also
/Case trnNrmDryRadFn_aal4b_02g - Det DRC - LimSurf - Response/
response
number some 1
region from 262 to 276
output to file also
/Case trnNrmDryRadFn_aal4b_02g - Det DRD - 1m - Response/
response
number some 1
region from 278 to 292
output to file also
/Case trnNrmDryRadFn_aal4b_02g - Det DRE - 1m+LimSurf - Response/
response
number some 1
region from 294 to 308
output to file also
end

*
* Unit 32 Material Specification
*
begin material specification
type dice
normalise
nmixtures 3
weight mixture 1
u235 3.2615E-02
u238 8.4888E-01
o 1.1850E-01
atoms mixture 2
h 6.6667E-01
o 3.3333E-01
atoms mixture 3
c 2.8571E-01
h 4.7619E-01
o 2.3810E-01

*
* Materials List - Dry Conditions - v1.2 - Class 1 - aal4b - STC Hybrid14 (Rev 0) Fuel
*
nmaterials 20
volume ! Homogenized aal4b Fuel
material 1
mixture 1 density 10.4120 prop 3.1489E-01 ! UO2 mixture at 3.7%
void prop 1.6671E-02 ! Gap
zircalloy density 6.5500 prop 9.7331E-02 ! Tube, clad
void prop 5.7111E-01 ! Interstitial, inside tubes
volume ! Fuel pin cladding
material 2
zircalloy density 6.5500 prop 1.0000
volume ! Water In Lattice and Tube
material 3
mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume ! Water In Fuel Rod Clad Gap
material 4
mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume ! Lower Nozzle Material

```


Figure 5.5-3 MCBEND Input File for Directly Loaded 14×14 Fuel Neutron Response from
Energy Group 2 – Normal Conditions (continued)

```

material 5
  stainless 3041 steel density 7.9200 prop 0.2948
  zircalloy density 6.5500 prop 0.0831
  void prop 0.6220
volume ! Upper Nozzle Material
material 6
  stainless 3041 steel density 7.9200 prop 0.3613
  void prop 0.6387
volume ! Upper Plenum Material
material 7
  stainless 3041 steel density 7.9200 prop 0.1147
  zircalloy density 6.5500 prop 0.0858
  void prop 0.7995
*
* Materials List - Common Materials - v1.2
*
volume ! Tube wall and cover sheet
material 8
  stainless 3041 steel density 7.9200 prop 1.0000
volume ! Structural Disk Material
material 9
  stainless 3041 steel density 7.9200 prop 1.0000
volume ! Weldment Material
material 10
  stainless 3041 steel density 7.9200 prop 1.0000
volume ! Heat Transfer Disk Material
material 11
  aluminium prop 1.0000
volume ! Canister Material
material 12
  stainless 3041 steel density 7.9200 prop 1.0000
atoms ! Transfer steel
material 13 density 0 ! (SCALE carbon steel)
  c prop 3.9250E-03
  fe prop 8.3498E-02
volume ! Lead
material 14
  pb density 11.0400 prop 1.0000
atoms ! NS-4-FR
material 15 density 0 ! 0 means atom/b-cm
  b10 prop 8.5500E-05
  b11 prop 3.4200E-04
  al prop 7.8000E-03
  h prop 5.8500E-02
  o prop 2.6100E-02
  c prop 2.2600E-02
  n prop 1.3900E-03
volume ! Stainless Steel 304
material 16
  stainless 3041 steel density 7.9200 prop 1.0000
volume ! Vent port middle cylinder
material 17
  stainless 3041 steel density 7.9200 prop 0.5000
  void prop 0.5000
volume ! Heat fins for transport cask
material 18
  cu density 8.9200 prop 0.4286
  stainless 3041 steel density 7.9200 prop 0.5714
volume ! Balsa
material 19
  mixture 3 density 0.1250 prop 1.0000
volume ! Redwood
material 20
  mixture 3 density 0.3870 prop 1.0000
end

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from Energy Group 7 – Accident Conditions

```

columns 1 200
*
* NAC-STC - aa14b_07g - Fuel Gamma - Radial
* Dry Cavity Conditions
* Accident Transport Conditions
* Transport Model Revision v1.6.2.0
* Shielding Revision v1.6
* STC Source Profile
* Cobalt Concentration of 1.2 g/kg
* Fuel Assembly Shift = None, Basket Shift = None
*
* Parameters
*
@samps = 10000000
*
* Unit 1 Control Data
*
begin control data
  run
  sample limit @samps
  time limit 1000m
  seeds 82896 43596
  chime every [@samps/10] samples
  report interim results
  sbd 30s
  dump intervals 1
end
*
* Unit 3 Output Control
*
*begin output control
* suppress inflows
*end
*
* Unit 4 Material Geometry
*
begin material geometry
* Fuel Assembly Type A - Class 1 - aa14b - STC Hybrid14 (Rev 0)
PART 1 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly Type B - Class 1 - aa14b - STC Hybrid14 (Rev 0)
PART 2 NEST
BOX M5 0.0000 0.0000 0.0000 19.7180 19.7180 8.6944 ! lower nozzle
BOX M1 S 0.0000 0.0000 0.0000 19.7180 19.7180 377.5024 ! fuel
BOX M7 0.0000 0.0000 0.0000 19.7180 19.7180 400.3040 ! top plenum
BOX M6 0.0000 0.0000 0.0000 19.7180 19.7180 409.1940 ! upper nozzle
* Fuel Assembly in Tube (Type A) v1.1
PART 3
BOX 1 1.4135 1.4135 0.0000 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of STC cavity
ZONES
/Fuel Assembly/ P1 +1
/Space in Tube/ H5 +2 -1
/Fuel Tube/ M8 +3 -2
/Container/ H5 +4 -3 -2 -1
VOLUMES UNITY
* Fuel Assembly in Tube (Type B) v1.1
PART 4

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

```

BOX 1 1.4135 1.4135 0.0000 19.7180 19.7180 409.1940 ! Fuel assembly
BOX 2 0.1219 0.1219 0.0000 22.3012 22.3012 419.1000 ! Space inside tube
BOX 3 0.0000 0.0000 6.3500 22.5450 22.5450 392.9380 ! Fuel tube
BOX 4 0.0000 0.0000 0.0000 22.5450 22.5450 419.1000 ! Container body - extent of STC cavity
ZONES
/Fuel Assembly/ P2 +1
/Space in Tube/ H5 +2 -1
/Fuel Tube/ M8 +3 -2
/Container/ H5 +4 -3 -2 -1
VOLUMES UNITY
* Type A Disk Opening with Tube v1.2
PART 5 CLUSTER
BOX P3 0.4547 0.4547 0.0000 22.5450 22.5450 419.1000 ! Fuel tube type A with fuel assy
BOX H5 0.0000 0.0000 0.0000 23.4544 23.4544 419.1000 ! Support disk opening width
* Type B Disk Opening with Tube v1.2
PART 6 CLUSTER
BOX P4 0.4547 0.4547 0.0000 22.5450 22.5450 419.1000 ! Fuel tube type B with fuel assy
BOX H5 0.0000 0.0000 0.0000 23.4544 23.4544 419.1000 ! Support disk opening width
* STC Basket v1.2
PART 7
BOX 1 -38.9153 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 1
BOX 2 -11.7272 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 2
BOX 3 15.4610 56.2432 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 3
BOX 4 -70.7136 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 4
BOX 5 -38.9153 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 5
BOX 6 -11.7272 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 6
BOX 7 15.4610 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 7
BOX 8 47.2592 29.0551 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 8
BOX 9 -70.7136 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 9
BOX 10 -38.9153 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 10
BOX 11 -11.7272 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 11
BOX 12 15.4610 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 12
BOX 13 47.2592 1.8669 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 13
BOX 14 -70.7136 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 14
BOX 15 -38.9153 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 15
BOX 16 -11.7272 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 16
BOX 17 15.4610 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 17
BOX 18 47.2592 -25.3213 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 18
BOX 19 -70.7136 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 19
BOX 20 -38.9153 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 20
BOX 21 -11.7272 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 21
BOX 22 15.4610 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 22
BOX 23 47.2592 -52.5094 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 23
BOX 24 -38.9153 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 24
BOX 25 -11.7272 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 25
BOX 26 15.4610 -79.6976 0.0000 23.4544 23.4544 419.1000 ! Basket Opening 26
ZROD 27 0.0000 0.0000 0.0000 89.9922 419.1000 ! Basket stack to cavity height
ZONES
/Opening01/ P5 +1
/Opening02/ P5 +2
/Opening03/ P5 +3
/Opening04/ P6 +4
/Opening05/ P5 +5
/Opening06/ P5 +6
/Opening07/ P5 +7
/Opening08/ P6 +8
/Opening09/ P5 +9
/Opening10/ P5 +10
/Opening11/ P5 +11
/Opening12/ P5 +12
/Opening13/ P5 +13
/Opening14/ P5 +14
/Opening15/ P5 +15
/Opening16/ P5 +16
/Opening17/ P5 +17
/Opening18/ P5 +18
/Opening19/ P6 +19
/Opening20/ P5 +20
/Opening21/ P5 +21
/Opening22/ P5 +22
/Opening23/ P6 +23

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14x14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

/Opening24/   P5   +24
/Opening25/   P5   +25
/Opening26/   P5   +26
/Basket/      H1   +27  -1  -2  -3  -4  -5
              -6  -7  -8  -9  -10 -11
              -12 -13 -14 -15 -16 -17
              -18 -19 -20 -21 -22 -23
              -24 -25 -26

VOLUMES      UNITY
* Basket in Cask Cavity v1.2
PART 8      NEST
ZROD P7      0.0000  0.0000  0.0000  89.9922  419.1000      ! Basket inserted - Includes gap to lid
ZROD H5      0.0000  0.0000  0.0000  90.1700  419.1000      ! Inserts flood matl to id of stc
* Transport Cask Inner Lid - With Ports v1.6.2.0
PART 9
ZROD 1      0.0000  0.0000  0.0000  100.3300  18.0848      ! Inner lid base
ZROD 2      0.0000  0.0000  18.0848  92.5957  4.7752      ! Inner lid cap
ZROD 3      0.0000  77.9907  0.0000  8.2931  22.8600      ! Drain port
ZROD 4      0.0000  -77.9907  0.0000  8.2931  22.8600      ! Vent port
ZROD 5      0.0000  0.0000  15.2400  85.6234  5.0800      ! Neutron shield
ZROD 6      0.0000  77.9907  15.2400  10.1600  5.0800      ! Cut circle 1 for neutron shield
ZROD 7      0.0000  -77.9907  15.2400  10.1600  5.0800      ! Cut circle 2 for neutron shield
BOX 8      -10.1600  77.9907  15.2400  20.3200  7.6327  5.0800      ! Cut box 1 for neutron shield
BOX 9      -10.1600  -85.6234  15.2400  20.3200  7.6327  5.0800      ! Cut box 2 for neutron shield
ZROD 10     0.0000  0.0000  0.0000  100.3300  22.8600      ! Container
ZONES
/Container/   M0   +10  -1  -2
/LidBase1/    M16  +1  -3  -4  -5  -6  -7
              -8  -9
/LidBase2/    M16  +1  +8  -6
/LidBase3/    M16  +1  +9  -7
/LidBase4/    M16  +1  +6  -3
/LidBase5/    M16  +1  +7  -4
/Nshield/     M15  +5  -6  -7  -8  -9
/LidCap1/     M16  +2  -3  -4  -5  -6  -7
              -8  -9
/LidCap2/     M16  +2  +8  -6
/LidCap3/     M16  +2  +9  -7
/LidCap4/     M16  +2  +6  -3
/LidCap5/     M16  +2  +7  -4
/DrainPort/   P10  +3
/VentPort/    P10  +4
VOLUMES      UNITY
* Transport Cask Inner Lid Port Model - With Covers v1.6.2.0
PART 10     CLUSTER
ZROD M0      0.0000  0.0000  0.0000  1.2700  10.8966      ! Bottom cylinder
ZROD M0      0.0000  0.0000  10.8966  4.1275  7.5184      ! Middle cylinder
ZROD M16     0.0000  0.0000  18.4150  8.2931  2.5400      ! Cover
ZROD M0      0.0000  0.0000  20.9550  8.2931  1.9050      ! Top cylinder
ZROD M16     0.0000  0.0000  0.0000  8.2931  22.8600      ! Inner lid material
* Transport Cask - Accident Conditions v1.6.2.0
PART 11
ZROD 1      0.0000  0.0000  -34.6710  110.1090  490.1184      ! Transport Cask
ZROD 2      0.0000  0.0000  0.0000  90.1700  419.1000      ! Cavity
ZROD 3      0.0000  0.0000  -20.8280  100.1776  5.0800      ! Bottom neutron shield
ZROD 4      0.0000  0.0000  419.1000  100.3300  22.8600      ! Inner lid
ZROD 5      0.0000  0.0000  0.0000  103.4034  408.9400      ! Lead shield cavity
ZROD 6      0.0000  0.0000  0.0000  95.2500  30.4800      ! Inner shell lower
ZCONE 7      0.0000  0.0000  30.4800  95.2500  93.9800  7.6200      ! Inner shell lower cone
ZROD 8      0.0000  0.0000  38.1000  93.9800  332.7400      ! Inner shell middle
ZCONE 9      0.0000  0.0000  370.8400  93.9800  95.2500  7.6200      ! Inner shell upper cone
ZROD 10     0.0000  0.0000  378.4600  95.2500  30.4800      ! Inner shell upper
ZP 11      402.9752      ! Top axial lead slump cut plane
ZP 12      5.9648      ! Bottom axial lead slump cut plane
YP 13      100.9322      ! Radial lead slump cut plane
ZROD 14     0.0000  0.0000  -3.8100  124.7140  410.5148      ! Radial neutron shield shell
ZROD 15     0.0000  0.0000  -2.6100  124.0790  408.1148      ! Radial neutron shield
ZP 16      2.4700      ! Insulation (void) cut plane
ZCONE 17     0.0000  0.0000  19.0500  124.7140  113.9190  24.1300      ! Top of rotating trunnion
BOX 18     -124.7140  -12.7000  -3.8100  249.4280  25.4000  22.8600      ! Bottom of rotating trunnion
BOX 19     -114.2238  -12.7000  -3.8100  228.4476  25.4000  22.8600      ! Bottom of rotating trunnion base
BOX 20     -124.7140  -7.6200  -3.8100  249.4280  15.2400  15.2400      ! Trunnion void box

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14x14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

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XROD 21 -124.7140 0.0000 11.4300 7.6200 249.4280 ! Trunnion void circle
BOX 22 -124.7140 -12.7000 -3.8100 249.4280 25.4000 46.9900 ! Trunnion extent box
ZSEC 23 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 14.6584 15.3416 ! Heat fin 1
ZSEC 24 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 29.6584 30.3416 ! Heat fin 2
ZSEC 25 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 44.6584 45.3416 ! Heat fin 3
ZSEC 26 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 59.6584 60.3416 ! Heat fin 4
ZSEC 27 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 74.6584 75.3416 ! Heat fin 5
ZSEC 28 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 89.6584 90.3416 ! Heat fin 6
ZSEC 29 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 104.6584 105.3416 ! Heat fin 7
ZSEC 30 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 119.6584 120.3416 ! Heat fin 8
ZSEC 31 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 134.6584 135.3416 ! Heat fin 9
ZSEC 32 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 149.6584 150.3416 ! Heat fin 10
ZSEC 33 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 164.6584 165.3416 ! Heat fin 11
ZSEC 34 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 179.6584 180.3416 ! Heat fin 12
ZSEC 35 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 194.6584 195.3416 ! Heat fin 13
ZSEC 36 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 209.6584 210.3416 ! Heat fin 14
ZSEC 37 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 224.6584 225.3416 ! Heat fin 15
ZSEC 38 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 239.6584 240.3416 ! Heat fin 16
ZSEC 39 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 254.6584 255.3416 ! Heat fin 17
ZSEC 40 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 269.6584 270.3416 ! Heat fin 18
ZSEC 41 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 284.6584 285.3416 ! Heat fin 19
ZSEC 42 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 299.6584 300.3416 ! Heat fin 20
ZSEC 43 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 314.6584 315.3416 ! Heat fin 21
ZSEC 44 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 329.6584 330.3416 ! Heat fin 22
ZSEC 45 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 344.6584 345.3416 ! Heat fin 23
ZSEC 46 0.0000 0.0000 -2.6100 110.1090 124.0790 408.1148 359.6584 360.3416 ! Heat fin 24
ZROD 47 0.0000 0.0000 -34.6710 124.7140 490.1184 ! Container
ZONES
/Cavity/ P8 +2
/OuterShell/ M16 +1 -2 -3 -4 -5
/InnerShell1/ M16 +6 -2
/InnerShell2/ M16 +7 -2
/InnerShell3/ M16 +8 -2
/InnerShell4/ M16 +9 -2
/InnerShell5/ M16 +10 -2
/InnerLid/ P9 +4
/BotNShield/ M15 +3
/LeadShield/ M14 +5 -6 -7 -8 -9 -10
+12 -11 -13
/TopSlumpVoid/ M0 +5 -10 +11
/BotSlumpVoid/ M0 +5 -6 -12
/RadVoid/ M0 +5 -11 +13 +12
/RadNShieldShell1/ M16 +14 -15 -1 -22
/RadNShieldShell2/ M16 +14 -15 -1 -22 -17 -18
/RadNShield1/ M21 +15 -1 -23 -24 -25 -26 -27 -28
-29 -30 -31 -32 -33 -34 -35 -36
-37 -38 -39 -40 -41 -42 -43 -44
-45 -46 +16 -22
/RadNShield2/ M21 +15 -1 +22 -17 -18 -34 -46
/InsulationVoid/ M0 +15 -1 -16 -22
/RotTrunUpper/ M16 +17 +22 +14 -1
/RotTrunLower/ M16 +19 +14 -1
/RotTrunSide/ M16 +18 +14 -19 -20 -21
/RotTrunBoxVoid/ M0 +20 +14 -19
/RotTrunCircVoid/ M0 +21 +14 -19 -20
/HeatFin1/ M18 +23
/HeatFin2/ M18 +24
/HeatFin3/ M18 +25
/HeatFin4/ M18 +26
/HeatFin5/ M18 +27
/HeatFin6/ M18 +28
/HeatFin7/ M18 +29
/HeatFin8/ M18 +30
/HeatFin9/ M18 +31
/HeatFin10/ M18 +32
/HeatFin11/ M18 +33
/HeatFin12/ M18 +34 -17 -18
/HeatFin13/ M18 +35
/HeatFin14/ M18 +36
/HeatFin15/ M18 +37
/HeatFin16/ M18 +38
/HeatFin17/ M18 +39

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Figure 5.5-4 MCBEND Input File for Directly Loaded 14x14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

/HeatFin18/  M18  +40
/HeatFin19/  M18  +41
/HeatFin20/  M18  +42
/HeatFin21/  M18  +43
/HeatFin22/  M18  +44
/HeatFin23/  M18  +45
/HeatFin24/  M18  +46      -17      -18
/Container/  M0   +47      -1       -14

VOLUMES UNITY
* Transport Cask Detector Description v1.6.2.0
PART      12
* Radial Detector DRA (Surface) Bodies
ZROD      1      0.0000  0.0000  -34.6710  124.7140  490.1184
ZROD      2      0.0000  0.0000  -34.6710  125.7140  32.6746
ZROD      3      0.0000  0.0000  -1.9964  125.7140  32.6746
ZROD      4      0.0000  0.0000  30.6781  125.7140  32.6746
ZROD      5      0.0000  0.0000  63.3527  125.7140  32.6746
ZROD      6      0.0000  0.0000  96.0272  125.7140  32.6746
ZROD      7      0.0000  0.0000  128.7018  125.7140  32.6746
ZROD      8      0.0000  0.0000  161.3764  125.7140  32.6746
ZROD      9      0.0000  0.0000  194.0509  125.7140  32.6746
ZROD     10      0.0000  0.0000  226.7255  125.7140  32.6746
ZROD     11      0.0000  0.0000  259.4000  125.7140  32.6746
ZROD     12      0.0000  0.0000  292.0746  125.7140  32.6746
ZROD     13      0.0000  0.0000  324.7492  125.7140  32.6746
ZROD     14      0.0000  0.0000  357.4237  125.7140  32.6746
ZROD     15      0.0000  0.0000  390.0983  125.7140  32.6746
ZROD     16      0.0000  0.0000  422.7728  125.7140  32.6746
* Radial Detector DRAA (SurfaceAzi) Bodies
ZROD     17      0.0000  0.0000  -35.6710  125.7140  492.1184
* Band 1 Bodies
ZSEC     18      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  250.0000  260.0000
ZSEC     19      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  260.0000  270.0000
ZSEC     20      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  270.0000  280.0000
ZSEC     21      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  280.0000  290.0000
ZSEC     22      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  290.0000  300.0000
ZSEC     23      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  300.0000  310.0000
ZSEC     24      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  310.0000  320.0000
ZSEC     25      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  320.0000  330.0000
ZSEC     26      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  330.0000  340.0000
ZSEC     27      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  340.0000  350.0000
ZSEC     28      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  350.0000  360.0000
ZSEC     29      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  0.0000  10.0000
ZSEC     30      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  10.0000  20.0000
ZSEC     31      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  20.0000  30.0000
ZSEC     32      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  30.0000  40.0000
ZSEC     33      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  40.0000  50.0000
ZSEC     34      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  50.0000  60.0000
ZSEC     35      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  60.0000  70.0000
ZSEC     36      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  70.0000  80.0000
ZSEC     37      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  80.0000  90.0000
ZSEC     38      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  90.0000  100.0000
ZSEC     39      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  100.0000  110.0000
ZSEC     40      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  110.0000  120.0000
ZSEC     41      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  120.0000  130.0000
ZSEC     42      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  130.0000  140.0000
ZSEC     43      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  140.0000  150.0000
ZSEC     44      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  150.0000  160.0000
ZSEC     45      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  160.0000  170.0000
ZSEC     46      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  170.0000  180.0000
ZSEC     47      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  180.0000  190.0000
ZSEC     48      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  190.0000  200.0000
ZSEC     49      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  200.0000  210.0000
ZSEC     50      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  210.0000  220.0000
ZSEC     51      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  220.0000  230.0000
ZSEC     52      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  230.0000  240.0000
ZSEC     53      0.0000  0.0000  178.0984  125.7140  126.7140  30.0000  240.0000  250.0000
* Radial Detector DRB (1m) Bodies
ZROD     54      0.0000  0.0000  -134.6710  210.1090  690.1184
ZROD     55      0.0000  0.0000  -134.6710  211.1090  34.5059
ZROD     56      0.0000  0.0000  -100.1651  211.1090  34.5059
ZROD     57      0.0000  0.0000  -65.6592  211.1090  34.5059

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14x14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

ZROD	58	0.0000	0.0000	-31.1532	211.1090	34.5059			
ZROD	59	0.0000	0.0000	3.3527	211.1090	34.5059			
ZROD	60	0.0000	0.0000	37.8586	211.1090	34.5059			
ZROD	61	0.0000	0.0000	72.3645	211.1090	34.5059			
ZROD	62	0.0000	0.0000	106.8704	211.1090	34.5059			
ZROD	63	0.0000	0.0000	141.3764	211.1090	34.5059			
ZROD	64	0.0000	0.0000	175.8823	211.1090	34.5059			
ZROD	65	0.0000	0.0000	210.3882	211.1090	34.5059			
ZROD	66	0.0000	0.0000	244.8941	211.1090	34.5059			
ZROD	67	0.0000	0.0000	279.4000	211.1090	34.5059			
ZROD	68	0.0000	0.0000	313.9060	211.1090	34.5059			
ZROD	69	0.0000	0.0000	348.4119	211.1090	34.5059			
ZROD	70	0.0000	0.0000	382.9178	211.1090	34.5059			
ZROD	71	0.0000	0.0000	417.4237	211.1090	34.5059			
ZROD	72	0.0000	0.0000	451.9296	211.1090	34.5059			
ZROD	73	0.0000	0.0000	486.4356	211.1090	34.5059			
ZROD	74	0.0000	0.0000	520.9415	211.1090	34.5059			
* Radial Detector DRBA (lmAzi) Bodies									
ZROD	75	0.0000	0.0000	-135.6710	211.1090	692.1184			
* Band 1 Bodies									
ZSEC	76	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	250.0000	260.0000
ZSEC	77	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	260.0000	270.0000
ZSEC	78	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	270.0000	280.0000
ZSEC	79	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	280.0000	290.0000
ZSEC	80	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	290.0000	300.0000
ZSEC	81	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	300.0000	310.0000
ZSEC	82	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	310.0000	320.0000
ZSEC	83	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	320.0000	330.0000
ZSEC	84	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	330.0000	340.0000
ZSEC	85	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	340.0000	350.0000
ZSEC	86	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	350.0000	360.0000
ZSEC	87	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	0.0000	10.0000
ZSEC	88	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	10.0000	20.0000
ZSEC	89	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	20.0000	30.0000
ZSEC	90	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	30.0000	40.0000
ZSEC	91	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	40.0000	50.0000
ZSEC	92	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	50.0000	60.0000
ZSEC	93	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	60.0000	70.0000
ZSEC	94	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	70.0000	80.0000
ZSEC	95	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	80.0000	90.0000
ZSEC	96	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	90.0000	100.0000
ZSEC	97	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	100.0000	110.0000
ZSEC	98	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	110.0000	120.0000
ZSEC	99	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	120.0000	130.0000
ZSEC	100	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	130.0000	140.0000
ZSEC	101	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	140.0000	150.0000
ZSEC	102	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	150.0000	160.0000
ZSEC	103	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	160.0000	170.0000
ZSEC	104	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	170.0000	180.0000
ZSEC	105	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	180.0000	190.0000
ZSEC	106	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	190.0000	200.0000
ZSEC	107	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	200.0000	210.0000
ZSEC	108	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	210.0000	220.0000
ZSEC	109	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	220.0000	230.0000
ZSEC	110	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	230.0000	240.0000
ZSEC	111	0.0000	0.0000	178.0984	211.1090	212.1090	30.0000	240.0000	250.0000
* World									
ZROD	112	0.0000	0.0000	-185.6710	261.1090	792.1184			
* External Void									
ZROD	113	0.0000	0.0000	-235.6710	311.1090	892.1184			
ZONES									
/TransportCask/	P11	+1							
* Detector DRA (Surface)									
/DRA01/	M0	+2	-1						
/DRA02/	M0	+3	-1						
/DRA03/	M0	+4	-1						
/DRA04/	M0	+5	-1						
/DRA05/	M0	+6	-1						
/DRA06/	M0	+7	-1						
/DRA07/	M0	+8	-1						
/DRA08/	M0	+9	-1						

Figure 5.5-4 MCBEND Input File for Directly Loaded 14x14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

/DRA09/ M0      +10    -1
/DRA10/ M0      +11    -1
/DRA11/ M0      +12    -1
/DRA12/ M0      +13    -1
/DRA13/ M0      +14    -1
/DRA14/ M0      +15    -1
/DRA15/ M0      +16    -1
/Void/   M0      +17    -1
                -2     -3     -4     -5     -6     -7
                -8     -9    -10    -11    -12    -13
                -14    -15    -16
* Detector DRAA (SurfaceAzi)
/DRAA0101/ M0      +18
/DRAA0102/ M0      +19
/DRAA0103/ M0      +20
/DRAA0104/ M0      +21
/DRAA0105/ M0      +22
/DRAA0106/ M0      +23
/DRAA0107/ M0      +24
/DRAA0108/ M0      +25
/DRAA0109/ M0      +26
/DRAA0110/ M0      +27
/DRAA0111/ M0      +28
/DRAA0112/ M0      +29
/DRAA0113/ M0      +30
/DRAA0114/ M0      +31
/DRAA0115/ M0      +32
/DRAA0116/ M0      +33
/DRAA0117/ M0      +34
/DRAA0118/ M0      +35
/DRAA0119/ M0      +36
/DRAA0120/ M0      +37
/DRAA0121/ M0      +38
/DRAA0122/ M0      +39
/DRAA0123/ M0      +40
/DRAA0124/ M0      +41
/DRAA0125/ M0      +42
/DRAA0126/ M0      +43
/DRAA0127/ M0      +44
/DRAA0128/ M0      +45
/DRAA0129/ M0      +46
/DRAA0130/ M0      +47
/DRAA0131/ M0      +48
/DRAA0132/ M0      +49
/DRAA0133/ M0      +50
/DRAA0134/ M0      +51
/DRAA0135/ M0      +52
/DRAA0136/ M0      +53
/Void/   M0      +54    -17
                -18    -19    -20    -21    -22    -23
                -24    -25    -26    -27    -28    -29
                -30    -31    -32    -33    -34    -35
                -36    -37    -38    -39    -40    -41
                -42    -43    -44    -45    -46    -47
                -48    -49    -50    -51    -52    -53
* Detector DRB (1m)
/DRB01/ M0      +55    -54
/DRB02/ M0      +56    -54
/DRB03/ M0      +57    -54
/DRB04/ M0      +58    -54
/DRB05/ M0      +59    -54
/DRB06/ M0      +60    -54
/DRB07/ M0      +61    -54
/DRB08/ M0      +62    -54
/DRB09/ M0      +63    -54
/DRB10/ M0      +64    -54
/DRB11/ M0      +65    -54
/DRB12/ M0      +66    -54
/DRB13/ M0      +67    -54
/DRB14/ M0      +68    -54
/DRB15/ M0      +69    -54
/DRB16/ M0      +70    -54

```


Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

```

/DRB17/  M0      +71      -54
/DRB18/  M0      +72      -54
/DRB19/  M0      +73      -54
/DRB20/  M0      +74      -54
/Void/    M0      +75      -54
          -55      -56      -57      -58      -59      -60
          -61      -62      -63      -64      -65      -66
          -67      -68      -69      -70      -71      -72
          -73
* Detector DRBA (1mAzi)
/DRBA0101/ M0      +76
/DRBA0102/ M0      +77
/DRBA0103/ M0      +78
/DRBA0104/ M0      +79
/DRBA0105/ M0      +80
/DRBA0106/ M0      +81
/DRBA0107/ M0      +82
/DRBA0108/ M0      +83
/DRBA0109/ M0      +84
/DRBA0110/ M0      +85
/DRBA0111/ M0      +86
/DRBA0112/ M0      +87
/DRBA0113/ M0      +88
/DRBA0114/ M0      +89
/DRBA0115/ M0      +90
/DRBA0116/ M0      +91
/DRBA0117/ M0      +92
/DRBA0118/ M0      +93
/DRBA0119/ M0      +94
/DRBA0120/ M0      +95
/DRBA0121/ M0      +96
/DRBA0122/ M0      +97
/DRBA0123/ M0      +98
/DRBA0124/ M0      +99
/DRBA0125/ M0     +100
/DRBA0126/ M0     +101
/DRBA0127/ M0     +102
/DRBA0128/ M0     +103
/DRBA0129/ M0     +104
/DRBA0130/ M0     +105
/DRBA0131/ M0     +106
/DRBA0132/ M0     +107
/DRBA0133/ M0     +108
/DRBA0134/ M0     +109
/DRBA0135/ M0     +110
/DRBA0136/ M0     +111
/Void/    M0     -112      -75
          -76      -77      -78      -79      -80      -81
          -82      -83      -84      -85      -86      -87
          -88      -89      -90      -91      -92      -93
          -94      -95      -96      -97      -98      -99
          -100     -101     -102     -103     -104     -105
          -106     -107     -108     -109     -110     -111
/ExtVoid/ M-2000  +113     -112
Volumes
          1.0     15*2.5706E+04      1.0     36*6.6085E+02      1.0     20*4.5662E+04
          1.0     36*1.1080E+03      1.0     1.0
end

*
* Unit 5 Splitting Geometry for Radial Detectors - Gamma
*
begin splitting geometry
  r   37   fill  0.0000
        n   10   89.9922
        n    1   93.9800
        n   16  103.4034
        n    3  110.1090
        n    5  124.0790
        n    1  124.7140
        n    1  129.7140

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

z    38    fill    -39.6710
      n    1    -34.6710
      n    2    -20.8280
      n    2    -15.7480
      n    3     0.0000
      n    2     8.6944
      n   13    377.5024
      n    2    400.3040
      n    3    409.1940
      n    1    419.1000
      n    3    434.3400
      n    2    439.4200
      n    3    455.4474
      n    1    460.4474

end

*
* Unit 6 - Source Geometry for Fuel Gamma
*
begin source geometry
r    10    fill    0.0000    n    10    89.9922
z    13
      8.6944    17.9146    27.1348    45.5752    82.4560    156.2176
      229.9792    266.8600    303.7408    340.6216    349.8418    359.0620
      368.2822    377.5024
end

*
* Unit 7
*
begin energy data
gamma dice
importance standard 22 groups
scoring as importance
simple source histogram weighting automatic
end

*
* Unit 8 Importance Map - Radial
*
begin importance map
calculate
targets 15
part 12
zones
  2  3  4  5  6
  7  8  9 10 11
 12 13 14 15 16
strengths
  1.0E+02    1.0E+02    5.0E+01    5.0E+01    5.0E+01
  1.0E+01    1.0E+01    1.0E+00    1.0E+01    1.0E+01
  5.0E+01    5.0E+01    5.0E+01    1.0E+02    1.0E+02
defer mixing
void density 0.10
track
! coupled source
! write gamma importances to 32
! write unformatted file to 31
! use method d
end

*
* Unit 9 Scoring Data - Radial
*

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

```

begin scoring data
  flux
  part 12
  from 2 to 16 ! DRA
  from 18 to 53 ! DRAA
  from 55 to 74 ! DRB
  from 76 to 111 ! DRBA
  responses sos ditto
  contributions to responses ditto
  ! score distribution for response
  ! weight distribution total
end

*
* Unit 10 Response Data
*

begin response data
* Scaled to mrem/hr
/ansi ans-6.1.1-1977 photon flux-dose conversion factors - mcnp table h.2 - mrem/
function pairs
1.5000E+01 1.3300E-02
1.4787E+01 1.3142E-02
1.4577E+01 1.2985E-02
1.4370E+01 1.2831E-02
1.4166E+01 1.2678E-02
1.3964E+01 1.2528E-02
1.3766E+01 1.2379E-02
1.3570E+01 1.2231E-02
1.3377E+01 1.2086E-02
1.3187E+01 1.1942E-02
1.3000E+01 1.1800E-02
1.2785E+01 1.1641E-02
1.2573E+01 1.1483E-02
1.2365E+01 1.1328E-02
1.2160E+01 1.1175E-02
1.1958E+01 1.1025E-02
1.1760E+01 1.0876E-02
1.1565E+01 1.0729E-02
1.1374E+01 1.0584E-02
1.1185E+01 1.0441E-02
1.1000E+01 1.0300E-02
1.0781E+01 1.0136E-02
1.0567E+01 9.9740E-03
1.0357E+01 9.8149E-03
1.0152E+01 9.6583E-03
9.9499E+00 9.5043E-03
9.7522E+00 9.3526E-03
9.5585E+00 9.2035E-03
9.3686E+00 9.0566E-03
9.1824E+00 8.9122E-03
9.0000E+00 8.7700E-03
8.8374E+00 8.6521E-03
8.6777E+00 8.5358E-03
8.5210E+00 8.4211E-03
8.3670E+00 8.3079E-03
8.2158E+00 8.1962E-03
8.0674E+00 8.0861E-03
7.9216E+00 7.9774E-03
7.7785E+00 7.8701E-03
7.6380E+00 7.7644E-03
7.5000E+00 7.6600E-03
7.4214E+00 7.6031E-03
7.3436E+00 7.5467E-03
7.2666E+00 7.4907E-03
7.1905E+00 7.4351E-03
7.1151E+00 7.3799E-03
7.0406E+00 7.3251E-03
6.9668E+00 7.2707E-03
6.8937E+00 7.2167E-03
6.8215E+00 7.1632E-03
6.7500E+00 7.1100E-03
6.6983E+00 7.0721E-03

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

6.6469E+00	7.0344E-03
6.5959E+00	6.9969E-03
6.5454E+00	6.9596E-03
6.4952E+00	6.9225E-03
6.4454E+00	6.8856E-03
6.3960E+00	6.8489E-03
6.3469E+00	6.8124E-03
6.2983E+00	6.7761E-03
6.2500E+00	6.7400E-03
6.1981E+00	6.7021E-03
6.1466E+00	6.6643E-03
6.0956E+00	6.6268E-03
6.0450E+00	6.5895E-03
5.9948E+00	6.5524E-03
5.9450E+00	6.5155E-03
5.8956E+00	6.4788E-03
5.8467E+00	6.4423E-03
5.7981E+00	6.4061E-03
5.7500E+00	6.3700E-03
5.6979E+00	6.3331E-03
5.6463E+00	6.2963E-03
5.5952E+00	6.2598E-03
5.5445E+00	6.2235E-03
5.4943E+00	6.1874E-03
5.4446E+00	6.1515E-03
5.3953E+00	6.1158E-03
5.3464E+00	6.0803E-03
5.2980E+00	6.0451E-03
5.2500E+00	6.0100E-03
5.2244E+00	5.9887E-03
5.1990E+00	5.9674E-03
5.1737E+00	5.9462E-03
5.1485E+00	5.9251E-03
5.1235E+00	5.9041E-03
5.0985E+00	5.8831E-03
5.0737E+00	5.8622E-03
5.0490E+00	5.8414E-03
5.0245E+00	5.8207E-03
5.0000E+00	5.8000E-03
4.9744E+00	5.7797E-03
4.9490E+00	5.7594E-03
4.9236E+00	5.7393E-03
4.8985E+00	5.7192E-03
4.8734E+00	5.6991E-03
4.8485E+00	5.6792E-03
4.8237E+00	5.6593E-03
4.7990E+00	5.6394E-03
4.7744E+00	5.6197E-03
4.7500E+00	5.6000E-03
4.6975E+00	5.5619E-03
4.6455E+00	5.5240E-03
4.5941E+00	5.4863E-03
4.5433E+00	5.4490E-03
4.4931E+00	5.4118E-03
4.4434E+00	5.3750E-03
4.3942E+00	5.3384E-03
4.3456E+00	5.3020E-03
4.2975E+00	5.2659E-03
4.2500E+00	5.2300E-03
4.1971E+00	5.1886E-03
4.1449E+00	5.1474E-03
4.0934E+00	5.1066E-03
4.0425E+00	5.0662E-03
3.9922E+00	5.0260E-03
3.9425E+00	4.9862E-03
3.8935E+00	4.9467E-03
3.8451E+00	4.9075E-03
3.7972E+00	4.8686E-03
3.7500E+00	4.8300E-03
3.6967E+00	4.7863E-03
3.6442E+00	4.7429E-03
3.5924E+00	4.7000E-03

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

3.5414E+00	4.6574E-03
3.4911E+00	4.6152E-03
3.4415E+00	4.5734E-03
3.3926E+00	4.5320E-03
3.3444E+00	4.4910E-03
3.2968E+00	4.4503E-03
3.2500E+00	4.4100E-03
3.2019E+00	4.3683E-03
3.1546E+00	4.3269E-03
3.1079E+00	4.2860E-03
3.0619E+00	4.2454E-03
3.0166E+00	4.2052E-03
2.9720E+00	4.1655E-03
2.9280E+00	4.1260E-03
2.8847E+00	4.0870E-03
2.8420E+00	4.0483E-03
2.8000E+00	4.0100E-03
2.7793E+00	3.9906E-03
2.7588E+00	3.9713E-03
2.7384E+00	3.9520E-03
2.7182E+00	3.9329E-03
2.6981E+00	3.9138E-03
2.6782E+00	3.8949E-03
2.6585E+00	3.8760E-03
2.6388E+00	3.8573E-03
2.6193E+00	3.8386E-03
2.6000E+00	3.8200E-03
2.5569E+00	3.7780E-03
2.5146E+00	3.7364E-03
2.4729E+00	3.6953E-03
2.4319E+00	3.6547E-03
2.3917E+00	3.6145E-03
2.3520E+00	3.5747E-03
2.3131E+00	3.5354E-03
2.2747E+00	3.4965E-03
2.2371E+00	3.4580E-03
2.2000E+00	3.4200E-03
2.1563E+00	3.3744E-03
2.1135E+00	3.3293E-03
2.0715E+00	3.2849E-03
2.0303E+00	3.2410E-03
1.9900E+00	3.1978E-03
1.9504E+00	3.1551E-03
1.9117E+00	3.1130E-03
1.8737E+00	3.0714E-03
1.8365E+00	3.0304E-03
1.8000E+00	2.9900E-03
1.7553E+00	2.9381E-03
1.7118E+00	2.8872E-03
1.6693E+00	2.8371E-03
1.6279E+00	2.7879E-03
1.5875E+00	2.7395E-03
1.5481E+00	2.6920E-03
1.5096E+00	2.6453E-03
1.4722E+00	2.5994E-03
1.4356E+00	2.5543E-03
1.4000E+00	2.5100E-03
1.3537E+00	2.4512E-03
1.3089E+00	2.3937E-03
1.2656E+00	2.3376E-03
1.2237E+00	2.2828E-03
1.1832E+00	2.2293E-03
1.1441E+00	2.1771E-03
1.1062E+00	2.1260E-03
1.0696E+00	2.0762E-03
1.0342E+00	2.0275E-03
1.0000E+00	1.9800E-03
9.7793E-01	1.9477E-03
9.5635E-01	1.9160E-03
9.3525E-01	1.8848E-03
9.1461E-01	1.8541E-03
8.9443E-01	1.8238E-03

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

8.7469E-01	1.7941E-03
8.5539E-01	1.7649E-03
8.3651E-01	1.7361E-03
8.1805E-01	1.7078E-03
8.0000E-01	1.6800E-03
7.8939E-01	1.6633E-03
7.7892E-01	1.6467E-03
7.6859E-01	1.6303E-03
7.5839E-01	1.6141E-03
7.4833E-01	1.5980E-03
7.3841E-01	1.5821E-03
7.2861E-01	1.5663E-03
7.1895E-01	1.5507E-03
7.0941E-01	1.5353E-03
7.0000E-01	1.5200E-03
6.9483E-01	1.5118E-03
6.8970E-01	1.5037E-03
6.8461E-01	1.4955E-03
6.7955E-01	1.4875E-03
6.7454E-01	1.4795E-03
6.6956E-01	1.4715E-03
6.6461E-01	1.4635E-03
6.5971E-01	1.4557E-03
6.5483E-01	1.4478E-03
6.5000E-01	1.4400E-03
6.4482E-01	1.4318E-03
6.3968E-01	1.4236E-03
6.3458E-01	1.4155E-03
6.2952E-01	1.4075E-03
6.2450E-01	1.3994E-03
6.1952E-01	1.3915E-03
6.1458E-01	1.3835E-03
6.0968E-01	1.3756E-03
6.0482E-01	1.3678E-03
6.0000E-01	1.3600E-03
5.9480E-01	1.3507E-03
5.8965E-01	1.3415E-03
5.8454E-01	1.3324E-03
5.7948E-01	1.3233E-03
5.7446E-01	1.3142E-03
5.6948E-01	1.3053E-03
5.6455E-01	1.2964E-03
5.5966E-01	1.2875E-03
5.5481E-01	1.2787E-03
5.5000E-01	1.2700E-03
5.4478E-01	1.2596E-03
5.3962E-01	1.2493E-03
5.3450E-01	1.2391E-03
5.2943E-01	1.2290E-03
5.2440E-01	1.2190E-03
5.1943E-01	1.2090E-03
5.1450E-01	1.1991E-03
5.0962E-01	1.1893E-03
5.0479E-01	1.1796E-03
5.0000E-01	1.1700E-03
4.9476E-01	1.1607E-03
4.8957E-01	1.1514E-03
4.8444E-01	1.1422E-03
4.7937E-01	1.1331E-03
4.7434E-01	1.1241E-03
4.6937E-01	1.1151E-03
4.6445E-01	1.1062E-03
4.5958E-01	1.0974E-03
4.5477E-01	1.0887E-03
4.5000E-01	1.0800E-03
4.4473E-01	1.0701E-03
4.3952E-01	1.0603E-03
4.3438E-01	1.0506E-03
4.2929E-01	1.0409E-03
4.2426E-01	1.0314E-03
4.1930E-01	1.0220E-03
4.1439E-01	1.0126E-03

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

4.0953E-01	1.0033E-03
4.0474E-01	9.9411E-04
4.0000E-01	9.8500E-04
3.9469E-01	9.7374E-04
3.8946E-01	9.6260E-04
3.8429E-01	9.5160E-04
3.7920E-01	9.4072E-04
3.7417E-01	9.2996E-04
3.6920E-01	9.1933E-04
3.6431E-01	9.0882E-04
3.5947E-01	8.9843E-04
3.5470E-01	8.8815E-04
3.5000E-01	8.7800E-04
3.4465E-01	8.6531E-04
3.3937E-01	8.5279E-04
3.3418E-01	8.4046E-04
3.2907E-01	8.2831E-04
3.2404E-01	8.1633E-04
3.1908E-01	8.0453E-04
3.1420E-01	7.9290E-04
3.0939E-01	7.8143E-04
3.0466E-01	7.7014E-04
3.0000E-01	7.5900E-04
2.9458E-01	7.4511E-04
2.8926E-01	7.3147E-04
2.8403E-01	7.1809E-04
2.7890E-01	7.0495E-04
2.7386E-01	6.9205E-04
2.6891E-01	6.7938E-04
2.6405E-01	6.6695E-04
2.5928E-01	6.5474E-04
2.5460E-01	6.4276E-04
2.5000E-01	6.3100E-04
2.4448E-01	6.1661E-04
2.3909E-01	6.0255E-04
2.3381E-01	5.8881E-04
2.2865E-01	5.7538E-04
2.2361E-01	5.6226E-04
2.1867E-01	5.4943E-04
2.1385E-01	5.3690E-04
2.0913E-01	5.2466E-04
2.0451E-01	5.1269E-04
2.0000E-01	5.0100E-04
1.9433E-01	4.8721E-04
1.8882E-01	4.7380E-04
1.8346E-01	4.6076E-04
1.7826E-01	4.4808E-04
1.7321E-01	4.3575E-04
1.6829E-01	4.2376E-04
1.6352E-01	4.1210E-04
1.5888E-01	4.0075E-04
1.5438E-01	3.8973E-04
1.5000E-01	3.7900E-04
1.4404E-01	3.6809E-04
1.3832E-01	3.5749E-04
1.3282E-01	3.4720E-04
1.2754E-01	3.3721E-04
1.2247E-01	3.2750E-04
1.1761E-01	3.1807E-04
1.1293E-01	3.0892E-04
1.0845E-01	3.0002E-04
1.0414E-01	2.9139E-04
1.0000E-01	2.8300E-04
9.6496E-02	2.8039E-04
9.3115E-02	2.7781E-04
8.9852E-02	2.7526E-04
8.6704E-02	2.7272E-04
8.3666E-02	2.7021E-04
8.0734E-02	2.6772E-04
7.7906E-02	2.6526E-04
7.5176E-02	2.6282E-04
7.2542E-02	2.6040E-04

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

7.0000E-02      2.5800E-04
6.7684E-02      2.6103E-04
6.5444E-02      2.6410E-04
6.3279E-02      2.6721E-04
6.1185E-02      2.7035E-04
5.9161E-02      2.7353E-04
5.7203E-02      2.7675E-04
5.5311E-02      2.8000E-04
5.3481E-02      2.8330E-04
5.1711E-02      2.8663E-04
5.0000E-02      2.9000E-04
4.7510E-02      3.1092E-04
4.5144E-02      3.3335E-04
4.2896E-02      3.5740E-04
4.0760E-02      3.8318E-04
3.8730E-02      4.1083E-04
3.6801E-02      4.4047E-04
3.4968E-02      4.7224E-04
3.3227E-02      5.0631E-04
3.1572E-02      5.4284E-04
3.0000E-02      5.8200E-04
2.6879E-02      7.0502E-04
2.4082E-02      8.5404E-04
2.1577E-02      1.0346E-03
1.9332E-02      1.2532E-03
1.7321E-02      1.5181E-03
1.5518E-02      1.8390E-03
1.3904E-02      2.2277E-03
1.2457E-02      2.6986E-03
1.1161E-02      3.2690E-03
1.0000E-02      3.9600E-03

end

*
* Unit 13 Hole Data
*
begin hole data
* STC Basket Hole Description v1.2
* Hole 1 General Basket Structure
PLATE
0 0 1
7
417.8300 0 ! Top of Basket
389.4976 -2 ! Top of Highest Support Disk
326.4316 -7 ! Resume support disk only
79.2376 -4 ! Start of support-heat disk region
17.4396 -6 ! Bottom of Lowest Support Disk
0.0000 -3 ! Bottom of Basket
0.0000 0 ! Basket Offset
0

* Hole 2 Top Weldment Disk - no structure above the weldment disk
RZMESH
2 ! number of radial points
85.5472
89.9922
3 ! number of axial intervals
389.4976 ! Top of diskstack
400.5580 ! Bottom of weldment
403.0980 ! Top of weldment plate
417.8300 ! Void to top of basket
0 0 ! Material below weldment
10 10 ! Plate Material
0 10 ! Flange
0 ! Outside material

* Hole 3 Bottom Weldment Disk - no structure in the weldment disk support
RZMESH
1 ! number of radial points
89.9922
1 ! number of axial intervals
3.8100

```


Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

6.3500          ! Coordinates inherited from PLATE Hole
10              ! Plate Material
0              ! Outside material

* Hole    4      Support disk and heat transfer disk stack
PLATE
origin    0    0    79.2376    ! Origin
0    0    1
4
cell    12.3597          ! Sets up a repeating lattice of cells
12.3597    0              ! flood matl
7.6086    0              ! water gap
6.0211    11             ! aluminium disk
1.2700    0              ! water gap
9              ! steel disk

* Hole    5      Flood material model
PLATE
0    0    1
1
417.8300    0              ! Above flooded region
0              ! Flooded region

* Hole    6      Support disk stack lower
PLATE
origin    0    0    17.4396    ! Origin
0    0    1
2
cell    12.3596          ! Sets up a repeating lattice of cells
12.3596    0              ! flood matl
1.2700    0              ! water gap
9              ! steel disk

* Hole    7      Support disk stack upper
PLATE
origin    0    0    326.4316    ! Origin
0    0    1
2
cell    12.3596          ! Sets up a repeating lattice of cells
12.3596    0              ! flood matl
1.2700    0              ! water gap
9              ! steel disk

end

*
* Unit 15 Source Strength - Fuel Gamma
*
* Class 1 - aa14b - STC Hybrid14 (Rev 0) - Fuel Gamma - Group 7 Reponse
begin source strength
component    6.9739E-06    ! 1/volume (1/1.4339E+05)
component    r    10*1.0
component    z
4.2500E-01    6.7500E-01    9.0000E-01    1.1000E+00    1.2000E+00    1.1500E+00
1.0750E+00    1.0000E+00    8.6500E-01    7.4000E-01    6.5000E-01    5.5000E-01
4.5000E-01
component    energy
6*0.0
1.0000E+00
15*0.0

end

*
* Unit 16 Simple Source Weights
*
*begin source weights
*
*end

*
* Unit 31 Tabular Output

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from Energy Group 7 – Accident Conditions (continued)

```

*
begin tabular output
  /Case trnAccDryRadFg_aal4b_07g - Det DRA - Surface - Response/
  response interim
  number some 1
  region from 118 to 132
  output to file also
  /Case trnAccDryRadFg_aal4b_07g - Det DRRA - SurfaceAzi - Response/
  response
  number some 1
  region from 134 to 169
  output to file also
  /Case trnAccDryRadFg_aal4b_07g - Det DRB - 1m - Response/
  response
  number some 1
  region from 171 to 190
  output to file also
  /Case trnAccDryRadFg_aal4b_07g - Det DRBA - 1mAzi - Response/
  response
  number some 1
  region from 192 to 227
  output to file also
end

*
* Unit 32 Material Specification
*
begin material specification
type gamma
normalise
nmixtures 3
weight mixture 1
      u235 3.2615E-02
      u238 8.4888E-01
      o 1.1850E-01
atoms mixture 2
      h 6.6667E-01
      o 3.3333E-01
atoms mixture 3
      c 2.8571E-01
      h 4.7619E-01
      o 2.3810E-01

*
* Materials List - Dry Conditions - vl.3 - Class 1 - aal4b - STC Hybrid14 (Rev 0) Fuel
*
nmaterials 21
volume ! Homogenized aal4b Fuel
material 1
  mixture 1 density 10.4120 prop 3.1489E-01 ! UO2 mixture at 3.7%
  void prop 1.6671E-02 ! Gap
  zircalloy density 6.5500 prop 9.7331E-02 ! Tube, clad
  void prop 5.7111E-01 ! Interstitial, inside tubes
volume ! Fuel pin cladding
material 2
  zircalloy density 6.5500 prop 1.0000
volume ! Water In Lattice and Tube
material 3
  mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume ! Water In Fuel Rod Clad Gap
material 4
  mixture 2 density 0.9982 prop 1.0000 ! mixH2O
volume ! Lower Nozzle Material
material 5
  stainless 3041 steel density 7.9200 prop 0.2948
  zircalloy density 6.5500 prop 0.0831
  void prop 0.6220
volume ! Upper Nozzle Material
material 6
  stainless 3041 steel density 7.9200 prop 0.3613
  void prop 0.6387
volume ! Upper Plenum Material
material 7

```

Figure 5.5-4 MCBEND Input File for Directly Loaded 14×14 Fuel Gamma Response from
Energy Group 7 – Accident Conditions (continued)

```

stainless 3041 steel      density  7.9200  prop  0.1147
zircalloy      density  6.5500  prop  0.0858
void           prop  0.7995
*
* Materials List - Common Materials - vl.3
*
volume          ! Tube wall and cover sheet
material 8
  stainless 3041 steel      density  7.9200  prop  1.0000
volume          ! Structural Disk Material
material 9
  stainless 3041 steel      density  7.9200  prop  1.0000
volume          ! Weldment Material
material 10
  stainless 3041 steel      density  7.9200  prop  1.0000
volume          ! Heat Transfer Disk Material
material 11
  aluminium          prop  1.0000
volume          ! Canister Material
material 12
  stainless 3041 steel      density  7.9200  prop  1.0000
atoms          ! Transfer steel
material 13          density  0          ! (SCALE carbon steel)

  c          prop  3.9250E-03
  fe          prop  8.3498E-02
volume          ! Lead
material 14
  pb          density  11.0400  prop  1.0000
atoms          ! NS-4-FR
material 15          density  0          ! 0 means atom/b-cm
  b10          prop  8.5500E-05
  b11          prop  3.4200E-04
  al          prop  7.8000E-03
  h          prop  5.8500E-02
  o          prop  2.6100E-02
  c          prop  2.2600E-02
  n          prop  1.3900E-03
volume          ! Stainless Steel 304
material 16
  stainless 3041 steel      density  7.9200  prop  1.0000
volume          ! Vent port middle cylinder
material 17
  stainless 3041 steel      density  7.9200  prop  0.5000
  void          prop  0.5000
volume          ! Heat fins for transport cask
material 18
  cu          density  8.9200  prop  0.4286
  stainless 3041 steel      density  7.9200  prop  0.5714
volume          ! Balsa
material 19
  mixture 3          density  0.1250  prop  1.0000
volume          ! Redwood
material 20
  mixture 3          density  0.3870  prop  1.0000
atoms          ! NS-4-FR @ fire conditions
material 21          density  0          ! 0 means atom/b-cm
  b10          prop  8.5500E-05
  b11          prop  3.4200E-04
  al          prop  7.8000E-03
  h          prop  5.8500E-02
  o          prop  2.6100E-02
  c          prop  2.2600E-02
  n          prop  1.3900E-03
end

```

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6.0 CRITICALITY EVALUATION

6.1 Discussion and Results

The NAC-STC is designed to safely transport spent fuel in two configurations. Fuel assemblies may be placed directly into a fuel basket installed in the cask cavity (directly loaded) or may be sealed in a transportable storage canister (canistered). In the directly loaded configuration, the NAC-STC can transport 26 PWR fuel assemblies. The design basis fuels for the directly loaded configuration are the Westinghouse, Exxon/ANF/SPC, Combustion Engineering and Framatome-Cogema PWR fuel assemblies described in Table 6.2-1. In the canistered configuration, the NAC-STC can transport up to 36 Yankee Class fuel assemblies or up to 26 Connecticut Yankee fuel assemblies. The canistered configuration containing Yankee Class fuel is referred to as the Yankee-MPC. The canistered configuration containing Connecticut Yankee fuel is referred to as the Connecticut Yankee MPC (CY-MPC). The Yankee Class fuel assemblies are described in Table 6.2-2. The Connecticut Yankee fuel assemblies are described in Table 6.2-4.

The NAC-STC can also transport canistered Greater Than Class C (GTCC) waste. Since the GTCC waste does not contain fissionable isotopes, a criticality evaluation is not required.

This chapter demonstrates that the NAC-STC with the design basis spent fuel meets the criticality requirements of 10 CFR 71 Sections 71.55 and 71.59 [1]. As demonstrated by the criticality analyses presented in Section 6.4 and summarized below, the NAC-STC remains subcritical under all conditions and is assigned a Criticality Safety Index (CSI) of 0 ($N = \infty$) in accordance with 10 CFR 71.59.

6.1.1 Directly Loaded Fuel

The NAC-STC is designed to transport 26 directly loaded PWR fuel assemblies with an initial enrichment up to 4.2 wt % ^{235}U , with the exception of fuel assemblies meeting the geometric constraints of the 17 x 17 Framatome-Cogema AFA design, which is limited to 4.5 wt % ^{235}U . Criticality control in the NAC-STC is achieved using a flux trap principle. Each of the basket tubes in the NAC-STC are surrounded by four BORAL or TalBor neutron absorber sheets which are held in place by steel cladding. The neutron absorber sheets have a minimum $0.02 \text{ g }^{10}\text{B}/\text{cm}^2$ loading. The spacing of the basket tubes is maintained by the steel support disks. These disks provide water gap spacings between tubes of 1.64 inch and 3.46 inch. When the cask is flooded

with water, fast neutrons leaking from the fuel assemblies are thermalized in the water gaps and are absorbed in the neutron absorber sheets before causing a fission in an adjacent fuel assembly.

The SCALE 4.3 CSAS25 (SCALE 4.3, Landers and Petrie, 1995) calculational sequence is used to perform the NAC-STC criticality analysis. This sequence includes KENO-Va (Petrie) Monte Carlo analysis to determine the NAC-STC effective neutron multiplication factor (k_{eff}) under normal and accident conditions. The 27 group neutron library is used in all calculations, including those used to evaluate the sensitivity of the package to a range of moderator density and center-to-center spacing. The principal characteristics of the directly loaded assemblies are shown in Table 6.2-1. The most reactive directly loaded fuel assembly is the Framatome-Cogema 17 x 17 having an enrichment of 4.5 wt % ^{235}U . The analyses yielded the following maximum results:

Normal Conditions:	$k_{eff} \pm \sigma$	k_s
Loading – Moderator inside and dry outside	0.92541 ± 0.00086	0.93948
Transport – Dry inside and moderator outside	0.44315 ± 0.00032	0.44379
Hypothetical Accident Conditions:		
Fully Moderated	0.93388 ± 0.00083	0.94794

A typical CSAS25 input and output for Framatome-Cogema fuel is shown in Figure 6.7-10. Conservatisms contained in these analyses included: (1) 75 percent of the specified minimum ^{10}B loading in the BORAL or TalBor neutron absorber material; (2) infinite array of casks in the X-Y plane; (3) infinite fuel length with no inclusion of end leakage effects (standard basket configuration); (4) no structural material present in the assembly; (5) no dissolved boron in the cask cavity or surrounding loading or storage area; (6) no credit taken for fuel burnup or for the buildup of fission product neutron absorbers; and (7) moderator in the pellet to fuel rod clad gap during accident evaluations.

6.1.2 Canistered Yankee Class Fuel

The NAC-STC may transport a transportable storage canister containing up to 36 design basis Yankee Class fuel assemblies. Criticality control in the canister basket is also achieved using the flux trap principle. The flux trap controls the reactivity in the interior of each of the three basket configurations. In the first of the configurations, all fuel tubes are separated by a flux trap that is formed by surrounding the tube with stainless steel support disks and four $0.01\text{g } ^{10}\text{B}/\text{cm}^2$ (minimum) areal density BORAL sheets, which are held in place by stainless steel covers. This configuration is referred to as "standard" in following text. In the other configurations, the size of

four fuel tubes (one outer tube in each quadrant of the basket) is increased to form either enlarged fuel tubes or screened damaged fuel cans. The BORAL sheets are removed from the outside of the enlarged tubes and damaged fuel cans to retain a similar width as that of a standard fuel tube. The damaged fuel cans contain Yankee Class fuel assemblies with up to 20 damaged or missing fuel rods. The remainder of the tubes have BORAL sheets on each of the four sides. The spacing of the basket tubes is maintained by the stainless steel support disks. These disks provide water gap spacing between tubes of 0.75, 0.81 or 0.875 inches, depending on the tube placement within the basket. When the cask is flooded with water, fast neutrons leaking from the fuel assemblies are thermalized in the water gaps and are absorbed in the BORAL sheets before causing a fission in an adjacent fuel assembly. This NAC-STC canistered basket can accommodate up to 36 Yankee Class Zircaloy-clad assemblies with a maximum initial enrichment of 4.03 wt % ^{235}U or 36 Yankee Class stainless steel-clad assemblies with a maximum initial enrichment of 4.97 wt % ^{235}U .

The Yankee-MPC may contain one or more Reconfigured Fuel Assemblies. The Reconfigured Fuel Assembly is designed to confine the Yankee Class spent fuel rods, or portions thereof, which are classified as failed fuel. The total number of full-length rods in a reconfigured fuel assembly is less than the number contained in a Yankee Class fuel assembly (maximum of 64 versus 256 rods). Consequently, the reactivity of the Reconfigured Fuel Assembly, even with the most reactive fuel rods, is less than the design basis fuel assembly used in criticality (see Section 6.4.3.1).

The SCALE 4.3 CSAS25 calculational sequence is used to perform the Yankee-MPC canistered fuel criticality analysis, based on the use of the most reactive Yankee Class fuel assembly. This sequence includes KENO-Va Monte Carlo analysis to determine the effective neutron multiplication factor (k_{eff}) under normal and accident conditions. The 27 group ENDF/B-IV neutron cross-section library is used in all calculations, including those used to evaluate the sensitivity of the package to a range of moderator density and center-to-center spacing. The most reactive Yankee Class fuel is the United Nuclear Type A. The principal characteristics of this assembly are shown in Table 6.2-2. Normal and accident conditions were evaluated as shown below. The wet loading condition results are shown for information only. In normal loading of canistered fuel into the NAC-STC transport cask, the canister will be dry inside and out. Fuel loading in the canister will take place in the transfer cask.

The analyses yielded the following maximum results for the standard and enlarged fuel tube configurations:

Yankee-MPC, Normal Transport	$k_{eff} \pm \sigma$	k_s
Loading - Moderator inside and dry outside	0.8761 ± 0.0007	0.8942
Transport - Dry Inside and moderator outside	0.4580 ± 0.0006	0.4760
Yankee-MPC, Hypothetical Accident		
Fully Moderated	0.8834 ± 0.0008	0.9014
Fully Moderated – Enlarged fuel tubes	0.9003 ± 0.0007	0.9183

The maximum bias and uncertainty adjusted reactivity for the basket containing the damaged fuel cans is 0.9388 for a hypothetical accident condition. Increased reactivity over the standard configuration is the result of a combining a higher reactivity damaged fuel payload, fuel assemblies with up to 20 fuel rods removed, with the removal of the BORAL absorber sheets from the damaged fuel can, increasing the system interaction.

Fully moderated includes water inside and outside of the cask, including the neutron shield region, and inside and outside of the fuel, including the fuel pellet and cladding gaps. Only the hypothetical accident condition is presented for the enlarged fuel tube case, since it represents the bounding configuration.

Conservatisms contained in these analyses included: (1) most reactive Yankee Class fuel assembly class with maximum U loading; (2) 75 percent of the specified minimum ^{10}B loading in the neutron absorber sheets; (3) infinite array of casks in the X-Y plane; (4) infinite fuel length with no inclusion of end leakage effects; (5) no structural material present in the assembly; (6) no dissolved boron in the cask cavity or surrounding loading or storage area; (7) no credit taken for fuel burnup or for the buildup of fission product neutron absorber sheets; and (8) moderator assumed in the gap between the pellet and fuel rod clad.

6.1.3 Canistered Connecticut Yankee Fuel

The NAC-STC may also transport a CY-MPC canister containing up to 26 Connecticut Yankee fuel assemblies. The criticality evaluation of the NAC-STC containing the CY-MPC is performed with the MONK8a [5] Monte Carlo Program for Nuclear Criticality Safety Analysis. This code employs the Monte Carlo technique in combination with JEF 2.2-based point energy neutron libraries to determine the effective neutron multiplication factor (k_{eff}). MONK8a, with the JEF 2.2 neutron cross-section libraries, is benchmarked by comparison to critical

experiments relevant to Light Water Reactor fuel in storage and transport casks as shown in Section 6.5.2. The NUREG/CR-6361 [6] method-based verification performed for MONK8a has established an upper subcritical limit as a function of system parameters. For the Connecticut Yankee canistered fuel, the upper subcritical limit is 0.9425 (Section 6.5.2).

Criticality control in the CY-MPC basket is achieved using geometric control of the fuel assemblies along with the flux trap principle. Each of the fuel tubes in the basket is surrounded by four BORAL or TalBor neutron absorber sheets with a core areal density of $0.02 \text{ g } ^{10}\text{B}/\text{cm}^2$ (minimum). The sheets are held in place by stainless steel cladding. The center-to-center spacing of the fuel tubes is maintained by the stainless steel support disks.

Two configurations of the CY-MPC basket are available for loading: the standard 26-assembly basket configuration, and a 24-assembly basket configuration where two of the basket openings are blocked. The 26-assembly basket is analyzed for Zircaloy-clad assemblies with an initial enrichment of up to 3.93 wt % ^{235}U and for stainless steel clad assemblies with an initial enrichment of up to 4.03 wt % ^{235}U . These stainless steel clad fuel assemblies may also be loaded in the 24-assembly basket. Westinghouse Vantage 5 Zircaloy-clad assemblies with an initial enrichment of up to 4.61 wt % ^{235}U must be loaded in the 24-assembly basket. There are 53 Westinghouse Vantage 5H fuel assemblies in the Connecticut Yankee spent fuel inventory.

The remaining inventory may be loaded in either of the basket configurations. Evaluation of the CY-MPC reactivity is performed using the transfer cask shield geometry and considers the normal and accident conditions of transport. The reactivity of the transfer cask loaded with fuel is assumed to accurately represent the reactivity of the NAC-STC loaded with the same fuel. Additional conservative conditions and assumptions considered include the most reactive Connecticut Yankee fuel assembly type, 75% of the specified minimum ^{10}B loading in the neutron absorber, no credit taken for fuel burnup or for the buildup of fission product neutron absorber; worst case mechanical basket configuration, optimum moderation, including moderation in the gap between the pellet and fuel rod clad, and an infinite three-dimensional array of casks. The most reactive Connecticut Yankee fuel loading occurs with the 24-assembly basket fully loaded with Zircaloy-clad fuel assemblies with a maximum enrichment of 4.61 wt % ^{235}U . The 24-assembly basket configuration loaded with the most reactive fuel bounds the most reactive 26-assembly basket loading.

The maximum effective neutron multiplication factor from this loading is 0.3715 under dry conditions and 0.9327 under the postulated transport accident conditions involving full

moderator intrusion. Including two standard deviations establishes a system reactivity threshold, $k_{\text{eff}} + 2\sigma$, of 0.9343, which is less than the subcritical limit of 0.9425. Consequently, the most reactive configuration of the canistered Connecticut Yankee fuel in the NAC-STC, containing the most reactive fuel assemblies in the most reactive configuration, is well below the regulatory criticality safety limit, including all biases and uncertainties under normal and accident conditions.

6.2 Package Fuel Loading

This section presents the physical descriptions of the fuel types to be loaded into the NAC-STC transport cask and describes the loading configurations of the fuel that are considered in the criticality safety evaluations. These configurations include directly loaded PWR fuel, canistered Yankee Class PWR fuel and canistered Connecticut Yankee PWR fuel.

Fuel assemblies with zero burnup are used in these analyses. The fresh fuel assumption is conservative because the fuel becomes less reactive as burnup increases. The criticality evaluations for the transport configurations for directly loaded fuel and for the Yankee-MPC configuration are performed assuming that all of the fuel assembly rods are in place. Consequently, to preclude a potential increase in reactivity due to empty fuel rod positions in a spent fuel assembly, any fuel rods removed from an assembly lattice must be replaced with solid rods fabricated from Zircaloy or stainless steel. A separate analysis of empty fuel rod positions is performed for the CY-MPC fuel configurations. Consequently, loading of Connecticut Yankee fuel assemblies with missing fuel rods is permitted. The loading of unenriched fuel assemblies is not evaluated and is not permitted in any NAC-STC fuel loading configuration.

Directly Loaded Fuel

The directly loaded fuel assembly characteristics are presented in Table 6.2-1. The cask analysis identified similar reactivity for the Westinghouse 17 x 17 OFA fuel assembly at 4.2 wt % ^{235}U and the Framatome-Cogema AFA 17 x 17 assembly at 4.5 wt % ^{235}U . As described in Section 6.4.2.1, the reactivity of these assemblies was higher than the reactivity of the remaining fuel assemblies evaluated for direct loading. To establish a bounding reactivity condition, the fuel characteristic envelope (i.e., fuel geometric parameters and fuel mass) of the Framatome-Cogema AFA 17 x 17 fuel assembly was expanded. The assembly with the expanded parameters, labeled the AFAM, is more reactive than the remaining directly loaded fuel assemblies and is used as a design basis fuel.

Canistered Yankee Class Fuel

The NAC-STC can safely transport up to 36 Yankee Class fuel assemblies loaded in the Yankee-MPC transportable storage canister. The most reactive design basis Yankee Class fuel is the United Nuclear Type A assembly, as described in Section 6.4.3.1. This assembly is used in the criticality calculations for the Yankee-MPC canistered configuration. The Yankee fuel classes to be transported in the Yankee-MPC canistered fuel configuration are presented in Table 6.2-2.

Design parameters of the Reconfigured Yankee Class assemblies are presented in Table 6.2-3. The Reconfigured Fuel Assembly is shown in Figures 6.2-1 and 6.2-2. The number of Yankee Class assemblies in the canister is limited by the total assembly weight of 30,600 pounds.

The NAC-MPC may also hold a recaged fuel assembly in any fuel loading position. A recaged fuel assembly consists of United Nuclear Yankee Class fuel rods placed in a Combustion Engineering fuel assembly lattice (skeleton). The impact on system reactivity of the recaged fuel assembly is documented in Section 6.4.3.1.

A limited number of Yankee Class fuel assemblies have fuel rods replaced with fill rods. The fill rods consist of hollow Zircaloy rods holding either Zircaloy or stainless steel slugs. The reactivity evaluation of fuel assemblies with these fill rods is presented in Section 6.4.3.6.

Certain fuel assemblies have enrichments nominally greater than the nominal maximum enrichments considered in the design basis analysis. These higher enrichments are considered in Section 6.4.3.1, and are shown not to be significant.

Canistered Connecticut Yankee Fuel

The fuel types to be transported in the CY-MPC canistered fuel configurations are presented in Table 6.2-4. Westinghouse Vantage 5H fuel assemblies must be loaded in the 24-assembly basket. The remaining fuel types may be loaded in either the 26-assembly or 24-assembly basket. In addition, loading of the damaged fuel cans and reconfigured fuel assemblies is physically restricted to the four oversized corner locations of the CY-MPC basket. Design parameters of the reconfigured Connecticut Yankee assemblies (CY reconfigured fuel assembly) are presented in Table 6.2-5.

Physical characteristics of the various Connecticut Yankee fuel assemblies to be accommodated are presented in Tables 1.2-4 and 6.2-4. Connecticut Yankee employed both Zircaloy and stainless steel clad 15 x 15 fuel assemblies having a cross-section as shown in Figure 6.2-3. Stainless steel clad fuel assemblies were originally utilized and were produced by Westinghouse, B&W, B&W (GUNF), Gulf General Atomic, and NUMEC. Westinghouse, Gulf General Atomic, NUMEC, and B&W also produced the Zircaloy-clad fuel assembly types.

Either CY-MPC basket configuration may contain CY reconfigured fuel assemblies. The CY reconfigured fuel assembly consists of a square 10 x 10 array of tubes, as shown in Figure 6.2-4, designed to confine and maintain geometric configuration of individual spent fuel rods, or

portions thereof, or fuel pellets, which may be classified as damaged fuel or fuel debris. The total equivalent fuel mass in each tube is restricted to that of one spent fuel rod. The individual tubes are supported by tie plates in the upper and lower end fittings. The tie plates are closed with 250-micron screens to preclude the release of gross particulate material.

The total number of full-length fuel rods (100) that can be placed in the CY reconfigured fuel assembly is much less than the number that are in the fuel assemblies (204). Consequently, the reactivity of the CY reconfigured fuel assembly, even with the most reactive fuel rods, is less than the design basis fuel assembly used in criticality evaluations.

Either CY-MPC basket configuration may also contain damaged fuel cans. The damaged fuel can may hold a complete fuel assembly, a lattice or a failed rod storage canister. The outer dimensions of these cans require they be loaded in one of the four corner positions. The damaged fuel can is designed such that it can be handled in the same manner as a fuel assembly.

Figure 6.2-1 Yankee Class Reconfigured Fuel Assembly

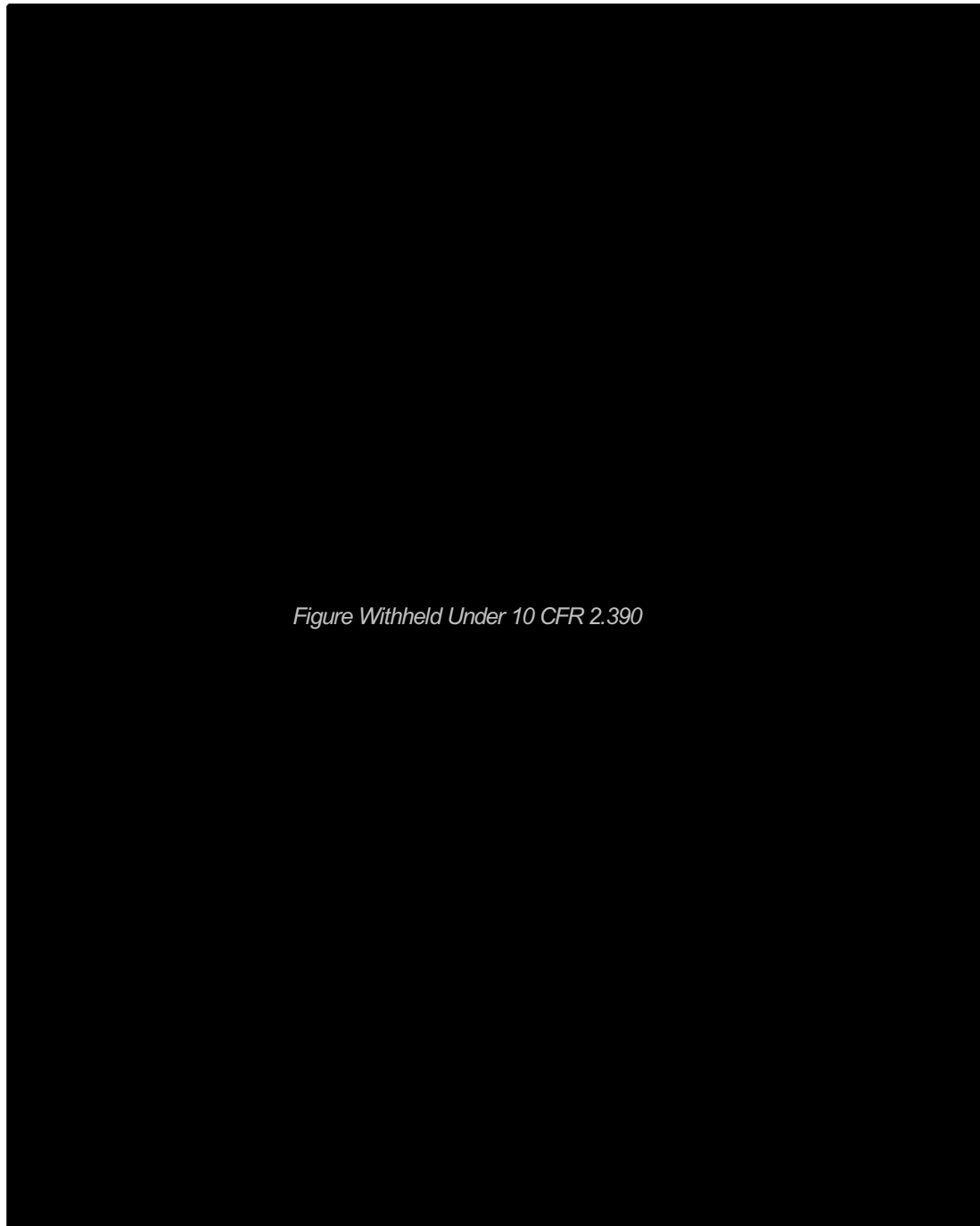


Figure Withheld Under 10 CFR 2.390

Figure 6.2-2 Yankee Class Reconfigured Fuel Assembly Cross-Section

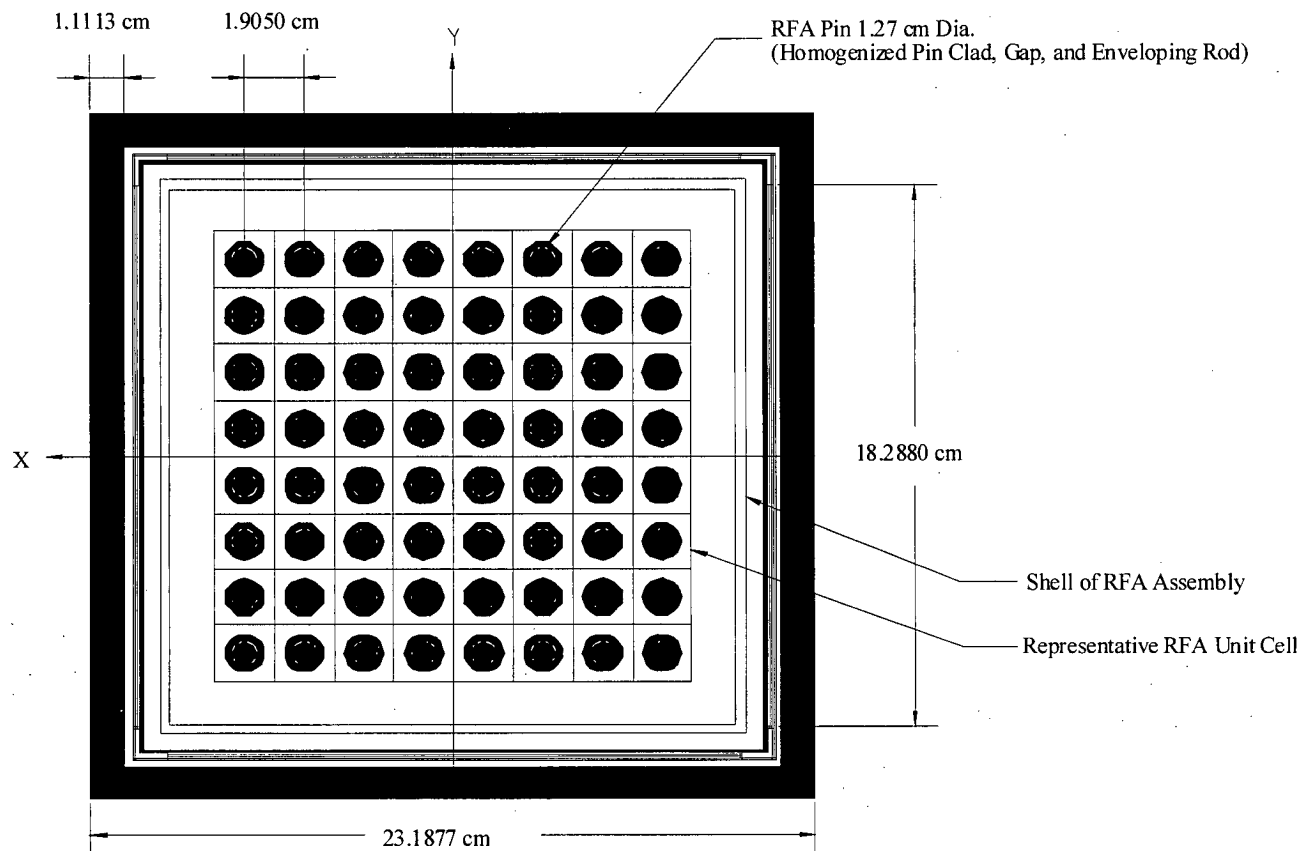


Figure 6.2-3 Connecticut Yankee 15 x 15 Fuel Assembly Array

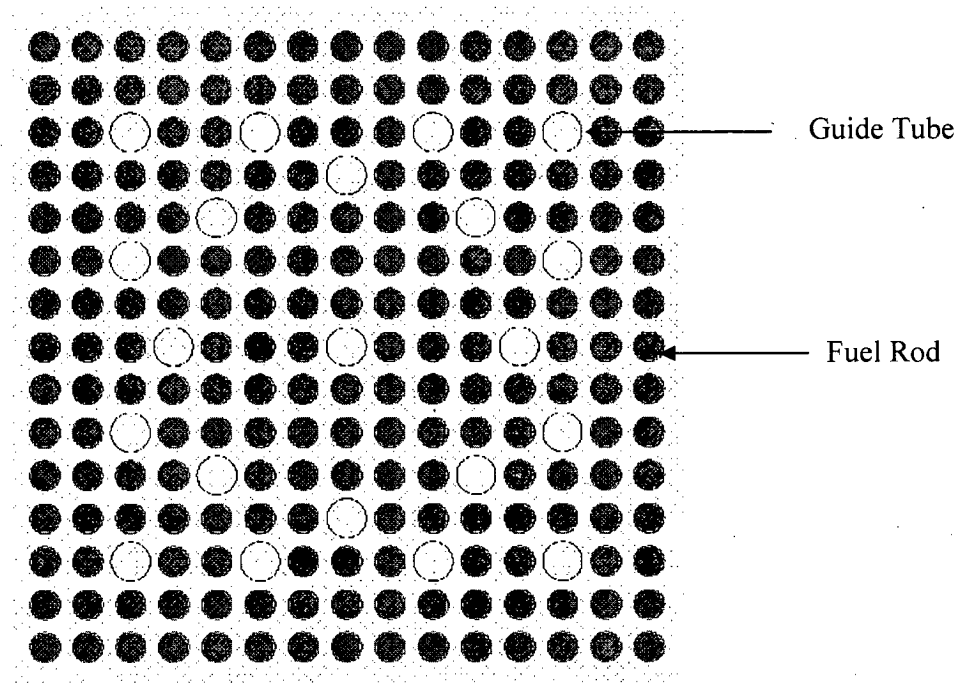


Figure 6.2-4 CY Reconfigured Fuel Assembly Array

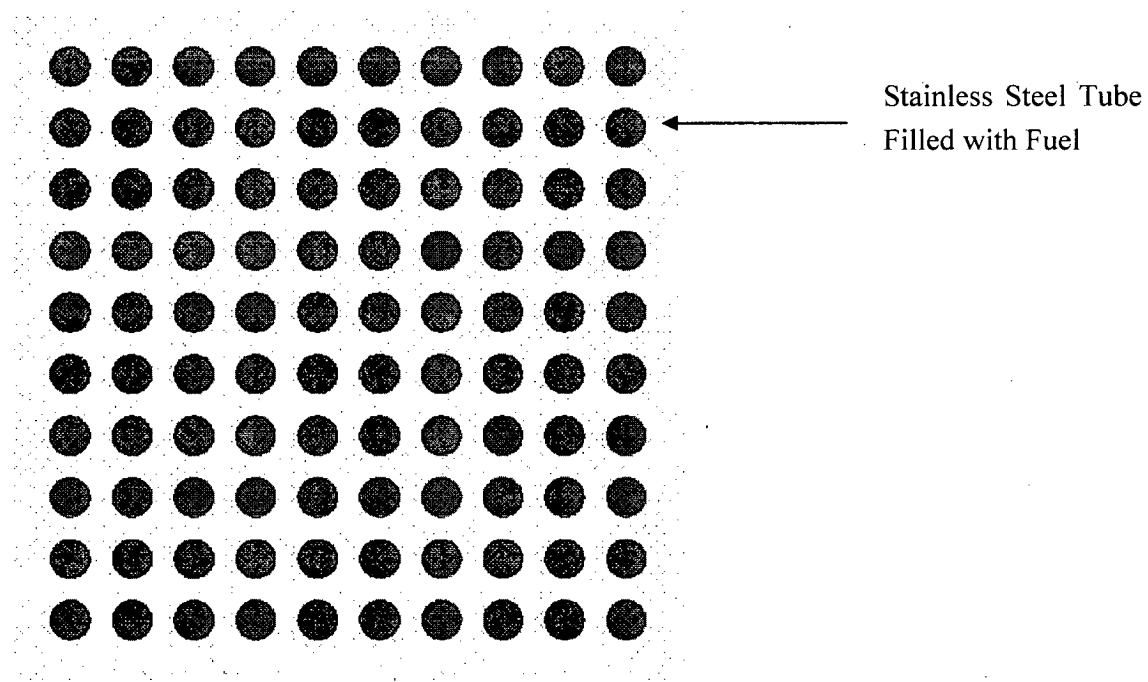


Table 6.2-1 Characteristics of Directly Loaded Fuel Assemblies

PWR Fuel Assembly Characteristics (Zirc-4 Clad) a Maximum Initial Enrichment of 4.2 wt % ²³⁵U

Vendor	Array	Version	ID	Enrichment (wt % ²³⁵ U)	Max MTU ¹	Number of Fuel Rods	Pitch (in)	Rod Dia. (in)	Clad Thick. (in)	Pellet Dia (in)	Active Length (in)
CE	14 X 14	Std.	14A1	4.2	0.4037	176	0.5800	0.4400	0.0280	0.3765	137.0
CE	14 X 14	Ft Cal.	14A2	4.2	0.3772	176	0.5800	0.4400	0.0280	0.3765	128.0
Ex/ANF	14 X 14	CE	14A3	4.2	0.3814	176	0.5800	0.4400	0.0310	0.3700	134.0
WE	14 X 14	CE Model	14A4	4.2	0.4115	176	0.5800	0.4400	0.0260	0.3805	136.7
Ex/ANF	14 X 14	WE	14B1	4.2	0.3689	179	0.5560	0.4240	0.0300	0.3505	142.0
Ex/ANF	14 X 14	Prairie Isl.	14B2	4.2	0.3741	179	0.5560	0.4170	0.0300	0.3505	144.0
WE	14 X 14	Std / ZCA	14B3	4.2	0.4144	179	0.5560	0.4220	0.0225	0.3674	145.2
WE	14 X 14	OFA	14B4	4.2	0.3612	179	0.5560	0.4000	0.0243	0.3444	144.0
WE	14 X 14	Std / ZCB	14B5	4.2	0.4144	179	0.5560	0.4220	0.0225	0.3674	145.2
Ex/ANF	15 X 15	WE	15A1	4.2	0.4410	204	0.5630	0.4240	0.0300	0.3565	144.0
WE	15 X 15	Std	15A2	4.2	0.4646	204	0.5630	0.4220	0.0242	0.3659	144.0
WE	15 X 15	Std / ZC	15A3	4.2	0.4646	204	0.5630	0.4220	0.0242	0.3659	144.0
WE	15 X 15	OFA	15A4	4.2	0.4646	204	0.5630	0.4220	0.0242	0.3659	144.0
CE	15 X 15	Palis.	15B1	4.2	0.4317	216	0.5500	0.4180	0.0260	0.3580	132.0
Ex/ANF	15 X 15	Palis	15B2	4.2	0.4310	216	0.5500	0.4170	0.0300	0.3580	131.8
CE	16 X 16	Lucie 2	16A1	4.2	0.4025	236	0.5060	0.3820	0.0250	0.3250	136.7
Ex/ANF	17 X 17	WE	17A1	4.2	0.4123	264	0.4960	0.3600	0.0250	0.3030	144.0
WE	17 X 17	Std	17A2	4.2	0.4671	264	0.4960	0.3740	0.0225	0.3225	144.0
WE	17 X 17	OFA	17A3	4.2	0.4282	264	0.4960	0.3600	0.0225	0.3088	144.0
WE	17 X 17	Vant 5	17A4	4.2	0.4282	264	0.4960	0.3600	0.0225	0.3088	144.0
FC	17 X 17	AFA	17A5	4.5	0.4669	264	0.4961	0.3740	0.0224	0.3224	144.0
FC	17 X 17	AFAM ²	17A6	4.5	0.4693	264	0.5011	0.3714	0.0204	0.3230	144.25

Notes:

- 1.) Based on 95% theoretical density and the listed fuel assembly dimensions.
- 2.) Represents the AFA fuel assembly with expanded fuel characteristics.

Table 6.2-2 Characteristics of Canistered Yankee Class Fuel Assemblies

Parameter	CE Type A	CE Type B	Exxon Type A	Exxon Type B	Exxon Type A	Exxon Type B	Westinghouse Type A	Westinghouse Type B	United Nuclear Type A	United Nuclear Type B
Assembly Configuration	-	-	-	-	-	-	-	-	-	-
Assembly Array	16x16	16x16	16x16	16x16	16x16	16x16	18x18	18x18	16x16	16x16
Max. Enrichment (wt % ²³⁵ U)	3.90	3.90	4.00	4.00	3.70	3.70	4.94	4.94	4.00	4.00
Max. MTU ¹	0.2394	0.2384	0.2394	0.2384	0.2394	0.2384	0.2869	0.2860	0.2456	0.2446
Fuel Rod Configuration	-	-	-	-	-	-	-	-	-	-
Fuel Rod Pitch (cm)	1.1989	1.1989	1.1989	1.1989	1.1989	1.1989	1.0719	1.0719	1.1887	1.1887
Active Fuel Length (cm)	231.1400	231.1400	231.1400	231.1400	231.1400	231.1400	233.9975	233.9975	231.1400	231.1400
Rod OD (cm)	0.9271	0.9271	0.9271	0.9271	0.9271	0.9271	0.8636	0.8636	0.9271	0.9271
Clad ID (cm)	0.8052	0.8052	0.8052	0.8052	0.8052	0.8052	0.7569	0.7569	0.8052	0.8052
Pellet OD (cm)	0.7887	0.7887	0.7887	0.7887	0.7887	0.7887	0.7468	0.7468	0.7887	0.7887
Diametral Gap (cm)	0.0165	0.0165	0.0165	0.0165	0.0165	0.0165	0.0102	0.0102	0.0165	0.0165
Rods per Assembly	231	230	231	230	231	230	305	304	237	236
Fuel Material	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂	UO ₂
Clad Material	Zircaloy	Zircaloy	Zircaloy	Zircaloy	Zircaloy	Zircaloy	SS 348	SS 348	Zircaloy	Zircaloy
Displacement Rod Configuration	-	-	-	-	-	-	-	-	-	-
Displacement Rod Material	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	Zircaloy - 4	Zircaloy - 4
Displacement Rod Diameter (cm)	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	0.9271	0.9271
Number Per Assembly	N/A	N/A	N/A	N/A	N/A	N/A	N/A	N/A	2	2
Guide Bar Configuration	-	-	-	-	-	-	-	-	-	-
Guide Bar Material	Zircaloy - 4	Zircaloy - 4	SS 304L	SS 304L	Zircaloy	Zircaloy	N/A	N/A	N/A	N/A
Guide Bar Width (cm)	1.0973	1.0973	1.0566	1.0566	1.0566	1.0566	N/A	N/A	N/A	N/A
Guide Bar Shape (cm)	Square	Square	Square	Square	Square	Square	N/A	N/A	N/A	N/A
Number Per Assembly	8	8	8	8	8	8	N/A	N/A	N/A	N/A
Instrument Tube Configuration	-	-	-	-	-	-	-	-	-	-
Instrument Tube ID (cm)	0.9970	0.9970	0.9970	0.9970	0.9970	0.9970	0.9995	0.9995	0.9995	0.9995
Instrument Tube OD (cm)	1.1481	1.1481	1.0884	1.0884	1.0884	1.0884	1.0884	1.0884	1.0884	1.0884
Number Per Assembly	1	1	1	1	1	1	1	1	1	1
Instrument Tube Material	Zircaloy - 4	Zircaloy - 4	SS 304	SS 304	Zircaloy	Zircaloy	SS 304	SS 304	SS 304	SS 304

1. Maximum MTU based on 95% of UO₂ theoretical density for the fuel pellet stack density.

Note: Certain fuel assemblies of the type specified may have a maximum enrichment up to 0.03 wt% ²³⁵U higher than the value shown.

Table 6.2-3 Yankee Class Reconfigured Fuel Assembly Parameters

Parameter	CE	Exxon	West.	United Nuclear
	Type A/B	Type A/B	Type A/B	Type A/B
ASSEMBLY CONFIGURATION				
Assembly Array	8x8	8x8	8x8	8x8
Max. Enrichment (wt % ²³⁵ U)	3.93	4.03	4.97	4.03
Max. kgU*	66.33	66.33	60.21	66.33
FUEL ROD CONFIGURATION (EACH ROD PLACED WITHIN ENCAPSULATING TUBE)				
Rod Pitch (cm)	1.905	1.905	1.905	1.905
Active Fuel Length (cm)	231.1400	231.1400	233.9975	231.1400
Rod OD (cm)	0.9271	0.9271	0.8636	0.9271
Clad ID (cm)	0.8052	0.8052	0.7569	0.8052
Pellet OD (cm)	0.7887	0.7887	0.7468	0.7887
Diametrical Gap (cm)	0.0165	0.0165	0.0102	0.0165
Max Rods per Assembly	64	64	64	64
Fuel Material	UO ₂	UO ₂	UO ₂	UO ₂
Clad Material	Zircaloy	Zircaloy	SS 348	Zircaloy
ENCAPSULATING TUBE				
Tube OD (cm)	1.27	1.27	1.27	1.27
Tube ID (cm)	1.1278	1.1278	1.1278	1.1278
Tube Material	SS-304	SS-304	SS-304	SS-304

* Maximum kgU based on 95% of UO₂ theoretical density for the fuel pellet stack density.

Table 6.2-4 Connecticut Yankee Design Basis Fuel Assembly Parameters

Vendor	WE	B&W	B&W, GGA, NUMEC	B&W (GUNF), GGA, NUMEC, WE
Array	15 X 15	15 X 15	15 X 15	15 X 15
Clad Material	Zircaloy	Zircaloy	Zircaloy	Stainless Steel
Maximum wt % ²³⁵ U	4.61	3.93	3.42	4.03
Maximum MTU ¹	0.3900	0.3742	0.3971	0.4337
Maximum Number of Fuel Rods	204	204	204	204
Maximum Pitch (in)	0.568	0.568	0.568	0.5702
Minimum Rod Diameter (in)	0.42	0.4194 ²	0.42	0.4175
Minimum Clad Thick (in)	0.0223	0.025	0.022	0.01375
Maximum Pellet Diameter (in)	0.3664	0.3615	0.3685	0.384
Maximum Active Length (in)	120.55	118.825	121.35	122.05
Minimum Guide Tube Thick (in)	0.013	0.008	0.008	0.008
Number Guide/Instrument Tubes	21	21	21	21
Guide/Instrument Tube Material	Zircaloy	Stainless Steel	Stainless Steel	Stainless Steel

1. Based on 95% of UO₂ theoretical density.
2. Results for a rod diameter of 0.42 in. in Section 6.4.2. Sensitivity analyses were performed to show that this rod diameter produced no statistically significant variation in the k_{eff} results.

Table 6.2-5 Connecticut Yankee Reconfigured Fuel Assembly Parameters

Parameter	Value
Array	10 x 10
Maximum Number of Fuel Rods	100
Maximum Enrichment (wt % ²³⁵ U)	4.61
Maximum kg U ¹	212
Maximum Tube Pitch (in)	0.81
Maximum Tube OD (in)	0.5675
Minimum Tube Thickness (in)	0.0315
Tube Material	SS 304

1. Based on 95% of UO₂ theoretical density.

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6.3 Criticality Model Specification

This section describes models used in the criticality evaluation of the directly loaded NAC-STC and the NAC-STC canistered fuel configurations. KENO-Va models are used in the criticality evaluation of the directly loaded NAC-STC and the Yankee-MPC canistered fuel configurations. The CY-MPC canistered fuel evaluation is performed using the MONK8a criticality code.

6.3.1 Calculational Methodology

The SCALE 4.3 CSAS25 calculational sequence is used to perform the NAC-STC criticality analysis for the directly loaded (uncanistered) and for canistered Yankee Class fuel. This sequence includes the SCALE Material Information Processor [7], BONAMI [8], NITAWL-II [9] and KENO-Va. The Material Information Processor generates number densities for standard compositions, prepares geometry data for resonance self-shielding, and creates data input files for the cross section processing codes. The BONAMI and NITAWL-II codes are used to prepare a resonance-corrected cross section library in AMPX working format. The KENO-Va code calculates the model k_{eff} using Monte Carlo techniques. The 27-group neutron library is used in the SCALE 4.5 criticality calculations. The validation of the CSAS25 sequence and the method statistics are addressed in Section 6.5.1. The KENO-Va models are described in Section 6.3.2.

The MONK8a Monte Carlo Program for Nuclear Criticality Safety Analysis is used to evaluate the CY-MPC system. This code employs the Monte Carlo technique in combination with JEF 2.2-based point energy neutron libraries for general neutron cross-section and thermal scatter in a water moderator to determine the effective neutron multiplication factor (k_{eff}). MONK8a, with the JEF 2.2 neutron cross-section libraries, is benchmarked by comparison to critical experiments relevant to Light Water Reactor fuel in storage and transport casks as shown in Section 6.5.2.

6.3.2 Description of Calculational Models

The NAC-STC KENO-Va model is derived from a radial slice of the cask at the central region. This section is the most reactive region due to the number of disks displacing water in the flux trap gap. The model is a stack of slices containing one aluminum disk, two identical water regions and one steel disk region (stack is aluminum, water, steel, water).

The cask body shielding regions of steel, lead, steel, NS-4-FR and steel surrounds each basket slice. Each cask slice is surrounded by a cuboid. The four slices are stacked into the KENO global unit. Periodic boundary conditions are imposed on the top and bottom to simulate an infinite cylinder, and reflecting boundary conditions are imposed on the sides simulating an infinite number of casks in the X-Y plane. Moderator density is varied both in the cask cavity regions normally filled with water and in the exterior cuboid.

For the directly loaded fuel configuration, the basket is modeled in each slice and contains 26 design basis fuel assemblies with a fuel density corresponding to 95% of the theoretical maximum. Enrichment varies from 4.2 wt % to 4.5 wt % ^{235}U . The fuel rod array is explicitly modeled in each of the 26 possible locations. For the Yankee-MPC canistered configuration, the basket model of each slice contains 36 Yankee Class design basis United Nuclear Type A fuel assemblies at a 4.0 wt % ^{235}U enrichment with a fuel density corresponding to 95% of the theoretical maximum. The fuel rod array is explicitly modeled in each of the 36 possible fuel locations.

The X-Y dimensions of the exterior cuboid are used to vary cask center-to-center spacing. Analysis of both normal and accident conditions use the same models except the models for accident conditions assume that the radial neutron shielding (NS-4-FR) is replaced by the external moderator. These models are shown in Figures 6.3-1 and 6.3-2.

An axially finite basket model is required for the damaged fuel can evaluation in which loose fuel may be located outside the neutron absorber sheet elevations. Details on the axially finite basket model are included in Section 6.4.3. Figure 6.3-5 depicts the location of the four loading positions for damaged fuel cans that also do not have BORAL sheets attached. These fuel cans are also modeled as simple square stainless steel boxes with an opening width of 7.99 inches and a wall thickness of 0.048 inch.

The MONK8a geometry modeling the CY-MPC is constructed as a finite model that accurately represents the geometry of the canister, the fuel and the basket. Figure 6.3-3 shows the cross-section of the structural disk that maintains separation of the fuel tubes in the cask. Also shown in the figure is a detail of the standard fuel tube that surrounds the fuel assembly and fits into the basket fuel position opening. Analyses of the CY-MPC normal and accident conditions are performed within the CY-MPC transfer cask shells. The geometry of these shells is shown in Figure 6.3-4. The MONK8a criticality code relies on a combinatorial logic geometry package. Therefore, intersecting simple geometry bodies are used to specify the fuel basket, canister, and

cask components. The appropriate value of each of the geometry bodies is modified to determine the worst case basket/cask condition within each analysis.

6.3.3 Package Regional Densities

The package regional densities used in the criticality analyses are:

<u>Material</u>	<u>Density (g/cc)</u>
UO ₂	10.41
Zircaloy	6.56
H ₂ O	0.9982
Steel	7.92
Lead	11.34
Aluminum	2.70
BORAL (core)	2.62
NS-4-FR	1.63

6.3.4 Fuel Region Densities

Fuel density corresponds to 95% of the theoretical density. Uranium isotope composition is provided for 4.0 and 4.5 wt % ²³⁵U fuel. The level of enrichment varies for individual categories of design basis directly loaded Yankee and Connecticut Yankee fuel.

<u>Material</u>	<u>Element</u>	<u>Density (atoms/barn-cm)</u>
UO ₂ (4.0 wt % ²³⁵ U)	²³⁵ U	9.406 x 10 ⁻⁴
	²³⁸ U	2.229 x 10 ⁻²
	O	4.646 x 10 ⁻²
UO ₂ (4.5 wt % ²³⁵ U)	²³⁵ U	1.058 x 10 ⁻³
	²³⁸ U	2.217 x 10 ⁻²
	O	4.646 x 10 ⁻²
Zircaloy	Zr	4.331 x 10 ⁻²
Stainless Steel		8.724 x 10 ⁻²
H ₂ O (0.9982 g/cm ³)	H	6.677 x 10 ⁻²
	O	3.338 x 10 ⁻²

6.3.5 Water Reflector Densities

The water reflector material densities outside the cask under normal operating conditions are:

<u>Material</u>	<u>Element</u>	Density (atom/barn-cm) <u>(directly loaded)</u>
H ₂ O	H	6.677×10^{-2}
	O	3.338×10^{-2}

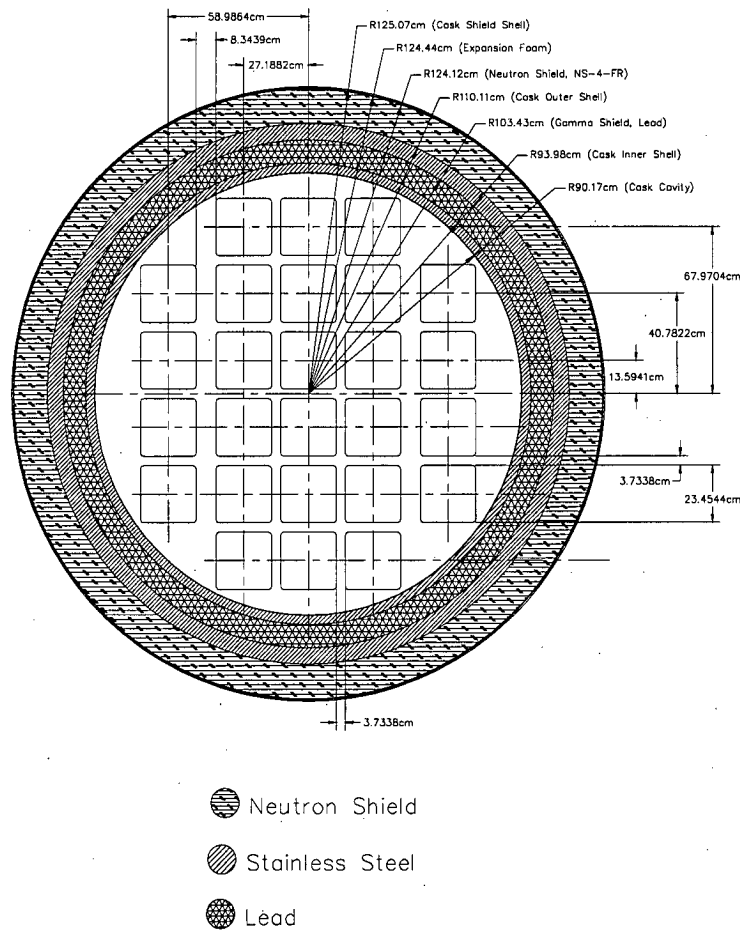
6.3.6 Cask Material Densities

The cask material densities for normal operating conditions are:

Material	Element	Density (atom/barn-cm) (directly loaded)	Density (atom/barn-cm) (canistered)
BORAL Core or TalBor ¹	¹⁰ B	7.098×10^{-3} (75% of Specified Minimum)	7.098×10^{-3} (75% of Specified Minimum)
	¹¹ B	3.925×10^{-2}	3.925×10^{-2}
	C	1.220×10^{-2}	1.220×10^{-2}
	Al	3.358×10^{-2}	3.358×10^{-2}
Aluminum	Al	6.031×10^{-2}	6.031×10^{-2}
Stainless Steel, Type 304	Cr	1.743×10^{-2}	1.743×10^{-2}
	Fe	5.936×10^{-2}	5.936×10^{-2}
	Ni	7.721×10^{-3}	7.721×10^{-3}
	Mn	1.736×10^{-3}	1.736×10^{-3}
Lead	Pb	3.297×10^{-2}	3.297×10^{-2}
NS-4-FR	H	5.854×10^{-2}	5.841×10^{-2}
	O	2.609×10^{-2}	2.607×10^{-2}
	C	2.264×10^{-2}	2.265×10^{-2}
	N	1.394×10^{-3}	1.401×10^{-3}
	Al	7.763×10^{-3}	7.781×10^{-3}
	¹¹ B	3.422×10^{-4}	3.565×10^{-4}
	¹⁰ B	8.553×10^{-5}	9.798×10^{-5}

1. TalBor is a 0.075-inch thick metal matrix material versus the core/clad BORAL material. The absorber density of TalBor is lower, but the required areal density is maintained.

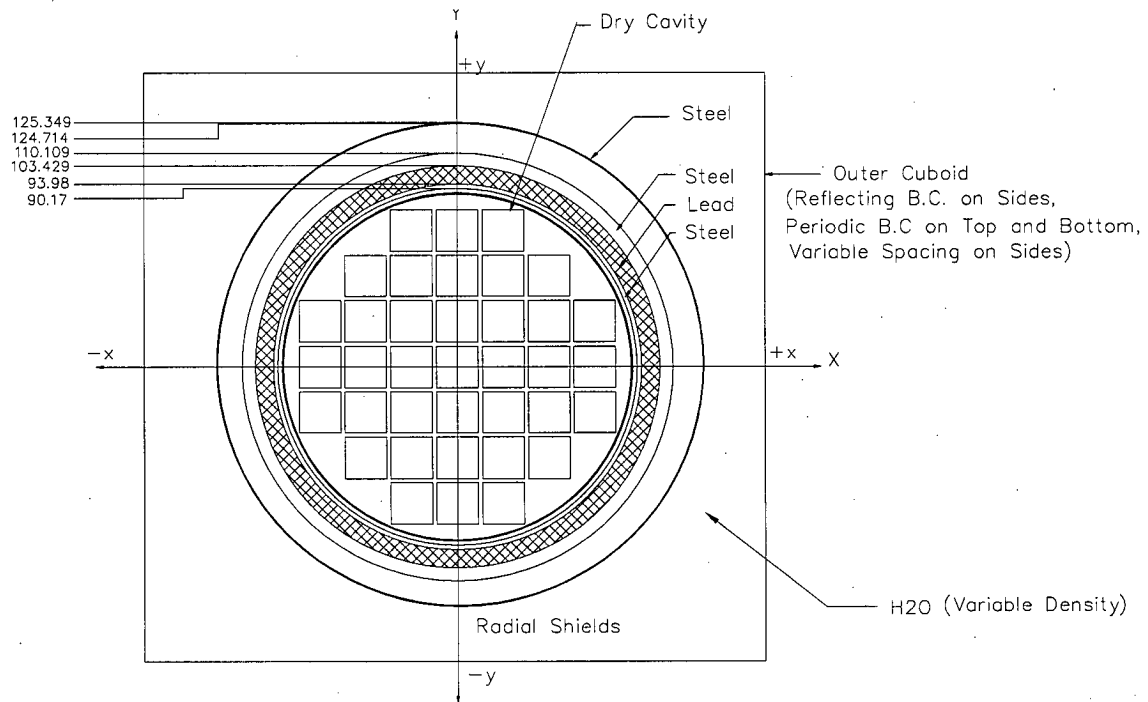
Figure 6.3-1 NAC-STC KENO-Va 26 Assembly Model - Directly Loaded Fuel Normal Conditions



Note: Disk Opening Coordinates are for Nominal Basket Configuration

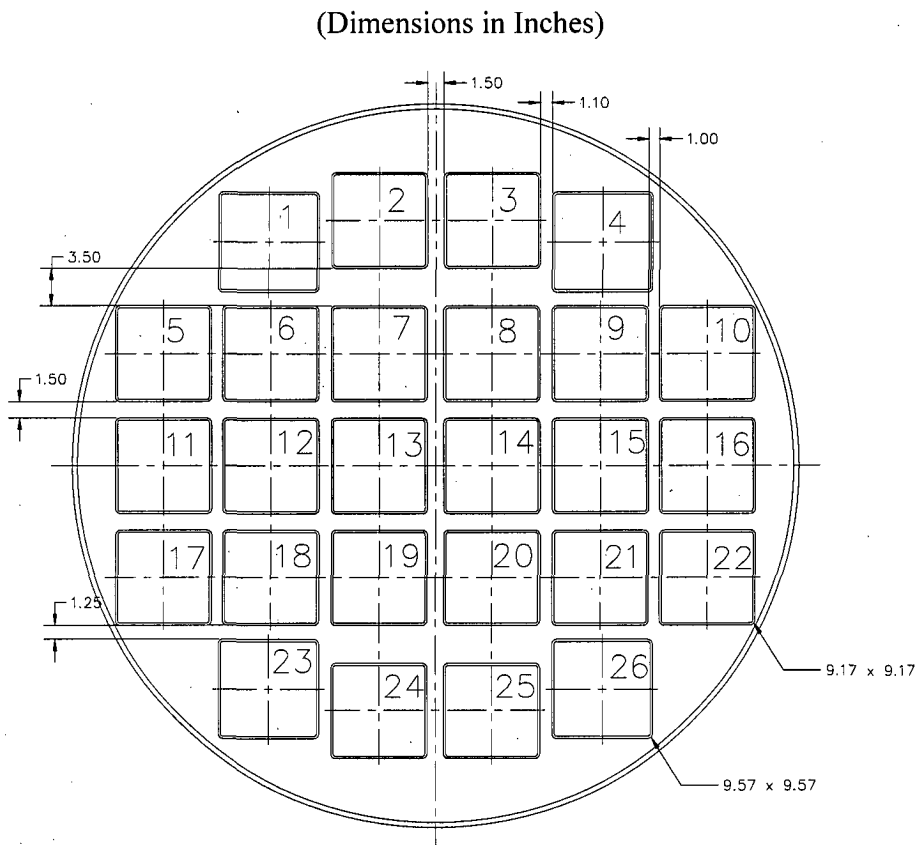
(All units are in centimeters)

Figure 6.3-2 NAC-STC KENO-Va 36 Assembly Model For Yankee-MPC Canistered Fuel



Dimensions in centimeters.

Figure 6.3-3 CY-MPC Basket Structural Disk and Fuel Tube Detail



Note: In the 24-assembly configuration, fuel positions numbered 12 and 15 are blocked.

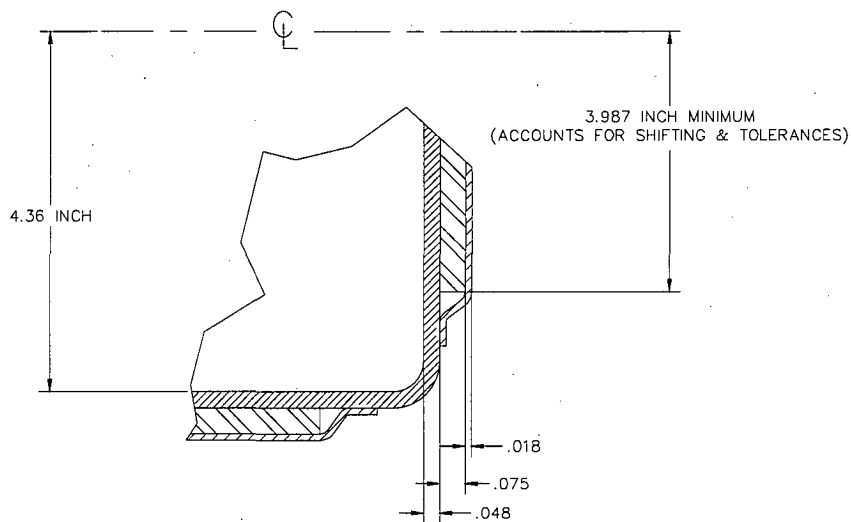


Figure 6.3-4 Transfer Cask Shells Used in the Evaluation of the CY-MPC

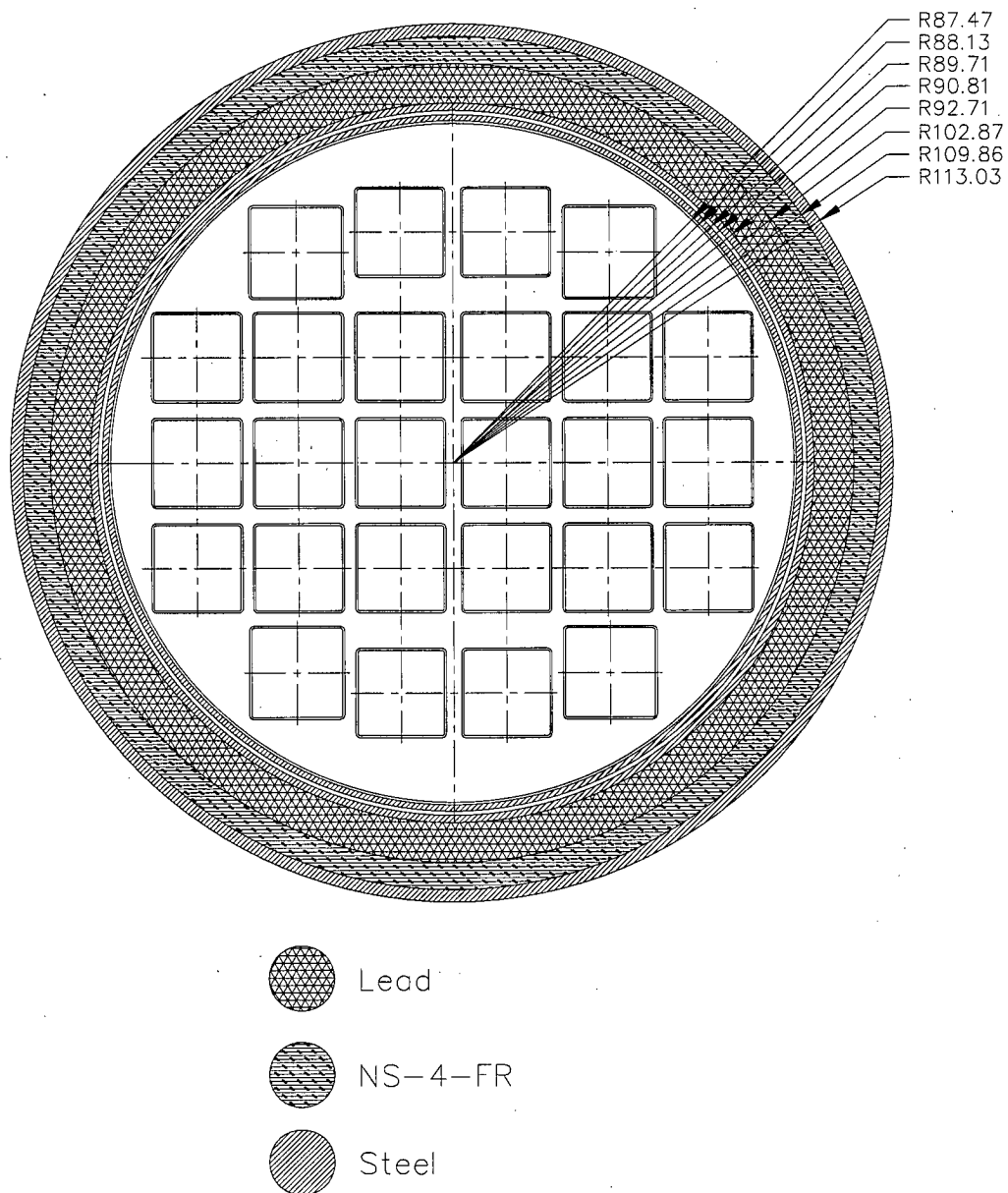
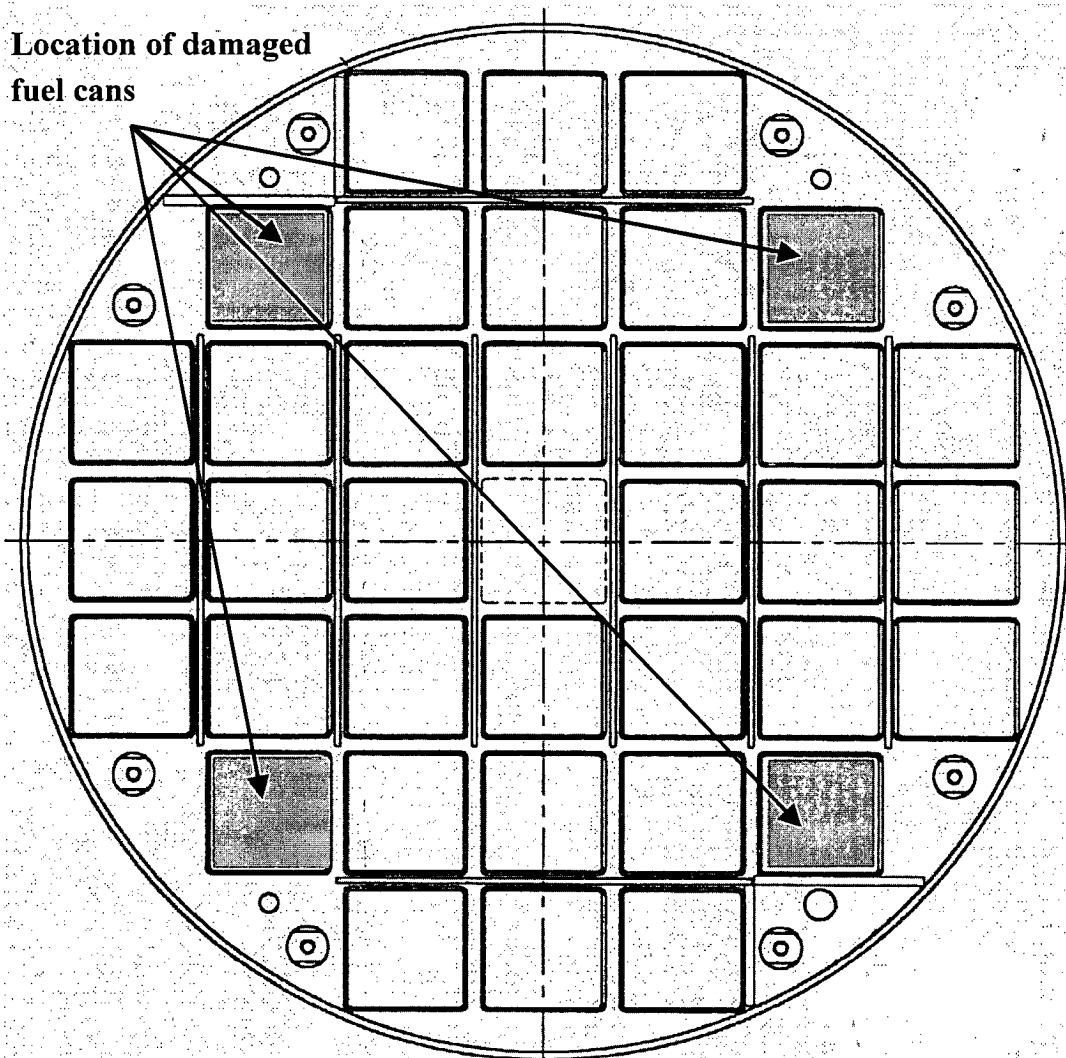


Figure 6.3-5 Damaged Fuel Can Locations



6.4 Criticality Calculation

This section demonstrates that the criticality safety of the NAC-STC, in either the directly loaded configuration, or the Yankee-MPC or CY-MPC canistered fuel configuration(s), satisfies the licensing requirements for the shipment of fissile material provided in 10 CFR 71.55 and 10 CFR 71.59.

10 CFR 71.55 and 10 CFR 71.59 require that the package remain subcritical under any credible condition, e.g., optimum interior/exterior moderation and reflection and credible configuration of the material. A criticality safety index is to be assigned to the fissile material package. This index reflects the number of packages (casks in this context) remaining subcritical in an array configuration.

Additional requirements imposed include the reduction in neutron absorber sheet ^{10}B from 100 to 75 percent and water in the pellet-to-cladding gap.

Undamaged Cask

Compliance with the requirements of paragraphs (b) and (d) of 10 CFR 71.55 is shown by modeling an undamaged cask surrounded by water. Requirements of paragraphs (a) through (c) of 10 CFR 71.59 are satisfied by providing a value of "N" equal to infinity and a CSI of 0 by imposing reflecting boundary conditions on the sides of the model simulating an infinite array of undamaged casks. Optimum interior and exterior moderation, including exterior full reflection by more than 20 cm of water, shows compliance with 10 CFR 71.55 paragraphs (b)(2), (b)(3) and (d)(3). Normal operating conditions for the canistered content transport cask include a dry canister cavity. The canister is loaded, dried, and seal welded inside a transfer cask. Only after the canister is dried and sealed is it placed into the transport cask. For conservatism, the interior of the canistered configurations is assumed to be fully flooded within the transport cask. This is identical to the analysis of loading operations for the directly loaded cask configuration. For canister loadings containing damaged fuel cans, preferential and uneven flooding conditions are evaluated. A representative set of exterior moderator density and cask pitch criticality evaluations shows compliance with 10 CFR 71 under dry cavity, transport conditions.

Damaged Cask

Compliance with the requirements of paragraph (e) of 10 CFR 71.55 is shown by modeling a damaged cask surrounded by water. Compliance with 10 CFR 71.59 is demonstrated by imposing reflection boundary conditions on the sides of the model to simulate an infinite array of damaged casks, thereby resulting in a CSI of 0. Optimum interior and exterior moderation, including exterior full reflection by more than 20 cm of water, shows compliance with 10 CFR 71.55 paragraphs (e)(2) and (e)(3) and 10 CFR 71.59 paragraph (a)(2).

A damaged transport cask is defined as having been subjected to the hypothetical accident conditions specified in 10 CFR 71. Under these conditions the cask containment is maintained, and the cavity, therefore, remains dry. However, to show the cask's capability to remain subcritical under optimum internal and external moderation, an internally flooded cask is analyzed. During the accident, the radial neutron shield is assumed to be lost as a result of fire and is replaced by the external moderator. Even though the fuel cladding is shown to remain intact following the cask drop, the pellet-to-clad gap is assumed to be filled with water. Introducing additional moderator into the normally under-moderated fuel assembly lattice increases reactivity.

6.4.1 Fuel Loading Optimization

The NAC-STC cask is designed to transport design basis PWR fuel assemblies in two (2) configurations, directly loaded fuel or canistered fuel. The criticality evaluation for directly loaded, uncanistered fuel is presented in Section 6.4.2. The analysis for canistered Yankee Class fuel in the Yankee-MPC is presented in Section 6.4.3. The analysis for canistered Connecticut Yankee fuel in the CY-MPC is presented in Section 6.4.4. These analyses illustrate that the maximum fuel loading, along with the most reactive configuration, is analyzed for each configuration. The configuration of fresh fuel in the cask, filled with water with no dissolved boron, and with the cask surrounded by water, is assumed to ensure that the maximum credible reactivity is simulated. Operationally, the canistered fuel configurations are loaded as previously sealed canisters, with both the cask and canister being in a dry condition.

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6.4.2 Criticality Results for Directly Loaded, Uncanistered Fuel

6.4.2.1 Most Reactive Assembly – Directly Loaded Fuel

A simplified KENO-Va calculation of the design basis assemblies for the directly loaded, uncanistered fuel condition described in Table 6.2-1, is performed to determine the most reactive fuel type. In this simplified model, a unit cell of the NAC-STC basket with the steel and aluminum webbing properly spaced axially is described. Reflecting boundary conditions are imposed on the sides, top and bottom simulating an infinite array of basket cells. All fuel assemblies are at the same fuel density, 95% of theoretical maximum. The k-infinity of the fuel assemblies in the NAC-STC basket are shown below. Also shown is the reactivity difference between the Westinghouse 17 x 17 OFA and the remaining evaluated assembly types. The difference is expressed as the ratio of the multiplication factor difference (Δk) and the Monte Carlo uncertainty.

Assembly	Enrichment wt % ^{235}U	k_{eff}	σ	$\Delta k/\sigma$
B&W 15x15 Mark B4	4.2	0.92051	0.00178	-5.09
B&W 17x17 Mark C	4.2	0.92371	0.00151	-3.88
CE 14x14	4.2	0.89363	0.00174	-20.66
CE 16x16 SYS 80	4.2	0.89376	0.00170	-21.06
West 14x14 Std	4.2	0.88147	0.00176	-27.33
West 14x14 OFA	4.2	0.89349	0.00180	-20.04
West 15x15	4.2	0.92326	0.00179	-3.53
West 17x17	4.2	0.91766	0.00180	-6.62
West 17x17 OFA	4.2	0.92957	0.00166	0.00
Exxon/ANF 14x14 CE	4.2	0.89413	0.00156	-22.72
Exxon/ANF 14x14 WE	4.2	0.87193	0.00169	-34.11
Exxon/ANF 15x15 WE	4.2	0.91629	0.00175	-7.59
Exxon/ANF 17x17 WE	4.2	0.92345	0.00172	-3.56
F-C AFA 17x17	4.2	0.91686	0.00171	-7.4
F-C AFA 17x17	4.5	0.93014	0.00163	0.3
F-C AFAM 17x17	4.2	0.92838	0.00185	-0.4
F-C AFAM 17x17	4.5	0.94089	0.00172	6.7

The most reactive fuel assemblies at 4.2 wt % ^{235}U are the Modified Framatome-Cogema AFA assembly (AFAM), and the Westinghouse 17 x 17 OFA. The standard 17 x 17 Westinghouse

and AFA fuels are significantly lower in reactivity. Maximum reactivity is obtained from the 4.5 wt % ^{235}U enriched Framatome-Cogema fuel. Specific evaluations for fuel enriched above 4.2 wt % ^{235}U are shown in Section 6.4.2.5.

Mechanical perturbation and moderator density studies are performed with the 4.2 wt % ^{235}U enriched Westinghouse 17 x 17 OFA. While enrichments over 4.2 wt. % ^{235}U are allowed for the AFA fuel types, the reactivity trends versus basket parameters, component movement, and moderator density are applicable to the higher enriched fuel. Modification to the enrichment level and the adjustment in fuel cross-section parameters will modify the magnitude of the reactivity change produced by the perturbation, but follow the same trend.

In particular, the H/U (moderator to fuel) ratio of the AFA and AFA modified fuels at 4.5 wt. % ^{235}U are below that of the Westinghouse 17x17 OFA assembly at 4.2 wt.% ^{235}U . Section 6.4.2.3 and 6.4.2.4 demonstrate that the Westinghouse fuel assembly is under-moderated. The AFA fuel assembly, with a lower H/U ratio, is therefore also under-moderated and does not require any additional moderator density studies. Increase in reactivity for the AFA modified assembly is associated with an increased fuel fissile material mass, compared to the Westinghouse standard and OFA 17x17 assemblies, in conjunction with an improved H/U ratio compared to the Westinghouse 17x17 standard assembly (but below that of the OFA assembly). The fissile material change has no impact on the relative flux trap behavior (i.e., tube movement) and will not impact the fuel assembly movement reactivity behavior (note that the fuel assembly is similar in size, < 0.09 inches wider, to the Westinghouse 17x17 assemblies). As shown in Section 6.4.2.2 the most reactive configuration concentrates fissile material in the center of the cask. The small change in fuel geometry (a slightly wider fuel assembly) has no impact on this behavior in that the fuel is still concentrated in the cask center and only the spacing between the center and the outer fuel assembly rings is slightly reduced (producing a higher reactivity system). Section 6.4.2.5 contains the criticality evaluation of AFAM fuel type at the most reactive system configuration.

6.4.2.2 Most Reactive Mechanical Configuration – Directly Loaded Fuel

Using the full cask model with the 4.2 wt % ^{235}U enriched Westinghouse 17 x 17 OFA fuel assembly, an evaluation of the effect of different directly loaded basket perturbations is made. This criticality analysis determines the most reactive basket mechanical configuration by altering the nominal model with the design basis assembly and comparing the perturbed k_{eff} to the nominal result. If Δk_{eff} ($k_{\text{perturbed}} - k_{\text{nominal}}$) is positive, the tolerance causes an increase in reactivity.

Conversely, if Δk_{eff} is negative, the tolerance causes a decrease in reactivity. To account for the statistical nature of the Monte Carlo analysis, and to determine if the change in reactivity is statistically significant, the Δk_{eff} is divided by the Monte Carlo uncertainty (σ) to arrive at a weight reactivity difference ($\Delta k_{\text{eff}}/\sigma$). Two sets of perturbations are assessed in the evaluation of criticality control: fabrication tolerances and component movement within the basket.

Four major fabrication tolerances are evaluated: 1) The fuel tube opening; 2) The disk opening; 3) The disk thickness; and, 4) The disk opening placement. The tolerances applied in the evaluation are ± 0.0762 cm for the tube opening, ± 0.0508 cm on the disk thickness, and ± 0.0381 cm on the disk opening size. The disk opening location tolerance is within a 0.0381 cm radius circle from the nominal location. The tolerance analysis results are:

Analysis	k_{eff}	σ	Δk_{eff}	$\Delta k_{\text{eff}}/\sigma$
Nominal Basket	0.90143	0.00090	----	----
Geometric Tolerances				
Min Tube	0.89494	0.00089	-0.00649	-7.292
Max Tube	0.90485	0.00085	0.00342	4.024
Min Disk Opening	0.89955	0.00087	-0.00188	-2.161
Max Disk Opening	0.90002	0.00086	-0.00141	-1.640
Shift Openings In	0.90169	0.00088	0.00026	0.295
Shift Openings Out	0.89799	0.00084	-0.00344	-4.095
Min Disk Thickness	0.89900	0.00087	-0.00243	-2.793
Max Disk Thickness	0.90073	0.00087	-0.00070	-0.805

Based on reactivity analysis, the only statistically significant change in reactivity occurs due to an increase in tube opening width. Increasing the fuel tube opening brings more moderator into the gap between the assembly and the tube lowering the efficiency of the neutron absorber sheets, hence increasing the reactivity of the system.

Two major component movements within the basket are evaluated: the assembly within the tube and the tube within the basket. Component movement is evaluated toward the top, right, top right, cask center, and cask periphery. Due to symmetry of the basket the remaining directions do not require analysis. To complete the analysis sequence, a combined radially inward shift of both fuel tube and assembly are evaluated.

As shown in the following table, based on the mechanical perturbation analysis, the maximum reactivity configuration of the basket is one in which both the fuel tube and fuel assembly are shifted toward the cask center.

Analysis	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff}/\sigma$
Nominal Basket	0.90143	0.00090	----	----
Mechanical Perturbations				
Assembly Shift Top Right	0.89811	0.00119	-0.00332	-2.790
Assembly Shift Top	0.89788	0.00122	-0.00355	-2.910
Assembly Shift Right	0.89763	0.00120	-0.00380	-3.167
Assembly Shift Radial In	0.90245	0.00130	0.00102	0.785
Assembly Shift Radial Out	0.89556	0.00119	-0.00587	-4.933
Fuel Tube Shift Top Right	0.89931	0.00124	-0.00212	-1.710
Fuel Tube Shift Top	0.90174	0.00118	0.00031	0.263
Fuel Tube Shift Right	0.89869	0.00121	-0.00274	-2.264
Fuel Tube Shift Radial In	0.90363	0.00126	0.00220	1.746
Fuel Tube Shift Radial Out	0.89361	0.00120	-0.00782	-6.517
Combined Analysis				
Tube + Assembly Radial In	0.90867	0.00120	0.00724	6.033

Thus, the following most reactive mechanical configuration is imposed on the NAC-STC directly loaded cask model: assemblies and fuel tubes moved toward the center of the basket, and maximum fuel tube opening.

6.4.2.3 Normal Conditions – Directly Loaded Fuel

Criticality results under normal conditions include variations in moderator density from 1.0 g/cc to 0.1 g/cc and cask center-to-center spacing from 250 cm (touching) to 300 cm. The results are shown in Tables 6.4.2-1 and 6.4.2-2. Table 6.4.2-1 shows the expected reactivity conditions during loading, i.e., wet inside and outside, as well as variation in moderator density due to draining and drying. Table 6.4.2-1 shows that cask reactivity is relatively insensitive to variations in cask center-to-center spacing. This results in a k_{eff} of 0.9129 ± 0.0009 . The CSAS25 input and output for this case is shown in Figure 6.7-1. Simultaneous variation in moderator density inside and outside the cask shows a monotonic decrease in reactivity. There appears to be no optimum reactivity at low density conditions. The maximum k_{eff} in the dry situation is 0.4929 ± 0.0013 , at a cask pitch of 300 cm.

Table 6.4.2-2 shows the expected reactivity conditions during normal transport, i.e., dry inside and wet outside. When the cask cavity is dry, k_{eff} of the package is very low and is insensitive to variations of moderator density outside and cask center-to-center spacing. The maximum k_{eff} for this situation is 0.4096 ± 0.0009 , at a cask pitch of 270 cm.

Including statistical and method uncertainties, all results for the normal condition are below the 0.95 NRC criticality safety limit. Thus, compliance with 10 CFR 71.55 (b) and (d), as well as 10 CFR 71.75 (a) is demonstrated.

6.4.2.4 Hypothetical Accident Conditions – Directly Loaded Fuel

Criticality results under hypothetical accident conditions include variations in exterior moderator density from 1.0 g/cc to 0.1 g/cc (dry), as well as cask center-to-center spacing from 250 cm (touching) to 300 cm. The results are shown in Table 6.4.2-3. Under accident conditions, moderator is allowed in the neutron shield region and outside the cask. The maximum k_{eff} for this situation is 0.9190 ± 0.0009 . With the cask cavity dry, the k_{eff} of the package is low and insensitive to moderator density and cask spacing variation. The CSAS25 input and output for this case is shown in Figure 6.7-2.

Including statistical and method uncertainties, all results for the accident condition are well below the 0.95 NRC criticality safety limit. Thus, compliance with 10 CFR 71.55 (e) and 10 CFR 71.75 (b) is demonstrated.

6.4.2.5 High Enrichment Evaluation, 4.5 wt% ^{235}U

As shown in Section 6.4.2.1, the maximum reactivity directly loaded fuel assemblies are the 4.5 wt. % ^{235}U enriched Framatome-Cogema 17x17 configurations identified as type AFA and AFAM. The AFA fuel type at 4.5 wt. % ^{235}U is similar in reactivity to that of the Westinghouse 17x17 OFA at 4.2 wt. % ^{235}U . The modified version of the Framatome-Cogema fuel assembly, labeled AFAM, raises the fissile mass and moderator to fuel ratio, both of which increase system reactivity. Increasing the pellet diameter and active fuel length raises the fissile material mass in the assembly. The moderator-to-fuel ratio is increased by reducing the fuel rod outer diameter and the fuel clad and guide tube thickness. To provide maximum directly loaded fuel assembly reactivities, the AFAM assembly is evaluated in the cask model at the worst-case configuration documented in Section 6.4.2.2. This configuration involves a shifted radial inward fuel assembly and fuel tube

with a maximum tolerance tube opening. Evaluations are performed at normal and accident conditions. Accident conditions involve flooding the pellet to clad gap and assume removal of the neutron shield. As documented in Sections 6.4.2.3 and 6.4.2.4, no statistically significant differences in reactivity occur as a function of cask spacing and exterior moderator density.

When flooding 100% of the pellet to clad gaps, in the under-moderated fuel assembly lattice during hypothetical accident condition, variations in reactivity may be seen due to changes in fuel pellet diameter (i.e., an increased pellet diameter displaces moderator and may result in a combined decrease in system reactivity). For the modified AFA assembly (AFAM) the majority of reactivity increase observed is the result of an increased fuel rod pitch. Modification to the pellet diameter, within the range expected from a standard PWR fuel assembly (± 0.0005 inch), does not produce a resolvable impact on system reactivity. Since the increased pellet diameter provides for a larger fissile mass in the typically dry pellet to clad gap configuration, the increased pellet diameter was retained for the flooded gap analysis.

Normal Conditions:	$k_{\text{eff}} \pm \sigma$	k_s
Loading – Moderator inside and dry outside	0.92541 ± 0.00086	0.93948
Transport – Dry inside and moderator outside	0.44315 ± 0.00032	0.44379
Hypothetical Accident Conditions:		
Fully Moderated	0.93388 ± 0.00083	0.94794

To satisfy 10 CFR 71.55(b)(3), an analysis of the reflection of the containment system (inner shell) by water is performed for a single cask. This evaluation resulted in k_{eff} values of 0.92473 for a single flooded intact cask fully water reflected and 0.92454 for a containment system fully water reflected. There is no statistically significant difference between the cases.

Table 6.4.2-1 Criticality Results for Normal Conditions of Direct Fuel Loading

Cask Pitch	H ₂ O Inside	H ₂ O Outside	Neutron Shield	¹⁰ B	k _{eff}	σ	k _s
250 cm	1.0	1.0	Yes	75 %	0.91291	0.00086	0.92698
270 cm	1.0	1.0	Yes	75 %	0.91137	0.00085	0.92543
300 cm	1.0	1.0	Yes	75 %	0.91086	0.00087	0.92493
250 cm	0.8	0.8	Yes	75 %	0.84595	0.00083	0.86001
270 cm	0.8	0.8	Yes	75 %	0.84564	0.00083	0.85970
300 cm	0.8	0.8	Yes	75 %	0.84631	0.00083	0.86037
250 cm	0.6	0.6	Yes	75 %	0.76900	0.00114	0.78319
270 cm	0.6	0.6	Yes	75 %	0.76642	0.00110	0.78059
300 cm	0.6	0.6	Yes	75 %	0.76671	0.00117	0.78092
250 cm	0.4	0.4	Yes	75 %	0.67331	0.00106	0.68746
270 cm	0.4	0.4	Yes	75 %	0.67276	0.00104	0.68691
300 cm	0.4	0.4	Yes	75 %	0.67441	0.00110	0.68858
250 cm	0.2	0.2	Yes	75 %	0.55708	0.00121	0.57131
270 cm	0.2	0.2	Yes	75 %	0.55593	0.00120	0.57015
300 cm	0.2	0.2	Yes	75 %	0.55529	0.00110	0.56946
250 cm	0.1	0.1	Yes	75 %	0.49153	0.00123	0.50577
270 cm	0.1	0.1	Yes	75 %	0.49294	0.00130	0.50722
300 cm	0.1	0.1	Yes	75 %	0.49293	0.00134	0.50723

Table 6.4.2-2 Criticality Results for Normal Conditions of Transport of Directly Loaded Fuel

Cask Pitch	H ₂ O Inside	H ₂ O Outside	Neutron Shield	¹⁰ B	k _{eff}	σ	k _s
250 cm	0.0001	1.0	Yes	75 %	0.40726	0.00084	0.42132
270 cm	0.0001	1.0	Yes	75 %	0.40776	0.00106	0.42191
300 cm	0.0001	1.0	Yes	75 %	0.40638	0.00086	0.42045
250 cm	0.0001	0.8	Yes	75 %	0.40775	0.00096	0.42186
270 cm	0.0001	0.8	Yes	75 %	0.40756	0.00092	0.42165
300 cm	0.0001	0.8	Yes	75 %	0.40704	0.00100	0.42117
250 cm	0.0001	0.6	Yes	75 %	0.40862	0.00085	0.42268
270 cm	0.0001	0.6	Yes	75 %	0.40788	0.00085	0.42194
300 cm	0.0001	0.6	Yes	75 %	0.40823	0.00081	0.42228
250 cm	0.0001	0.4	Yes	75 %	0.40805	0.00091	0.42214
270 cm	0.0001	0.4	Yes	75 %	0.40706	0.00080	0.42111
300 cm	0.0001	0.4	Yes	75 %	0.40580	0.00091	0.41989
250 cm	0.0001	0.2	Yes	75 %	0.40931	0.00092	0.42340
270 cm	0.0001	0.2	Yes	75 %	0.40933	0.00098	0.42345
300 cm	0.0001	0.2	Yes	75 %	0.40683	0.00082	0.42088
250 cm	0.0001	0.1	Yes	75 %	0.40663	0.00085	0.42069
270 cm	0.0001	0.1	Yes	75 %	0.40955	0.00094	0.42365
300 cm	0.0001	0.1	Yes	75 %	0.40796	0.00091	0.42205

Table 6.4.2-3 Criticality Results for Directly Loaded Fuel in Hypothetical Accident Conditions

Cask Pitch	H ₂ O Inside	H ₂ O Outside	Neutron Shield	¹⁰ B	k _{eff}	σ	k _s
250 cm	1.0	1.0	No	75 %	0.91902	0.00085	0.93308
270 cm	1.0	1.0	No	75 %	0.91787	0.00086	0.93194
300 cm	1.0	1.0	No	75 %	0.91799	0.00087	0.93206
250 cm	0.8	0.8	No	75 %	0.85275	0.00084	0.86681
270 cm	0.8	0.8	No	75 %	0.85247	0.00085	0.86653
300 cm	0.8	0.8	No	75 %	0.85157	0.00087	0.86564
250 cm	0.6	0.6	No	75 %	0.77755	0.00084	0.79161
270 cm	0.6	0.6	No	75 %	0.77531	0.00084	0.78937
300 cm	0.6	0.6	No	75 %	0.77623	0.00083	0.79029
250 cm	0.4	0.4	No	75 %	0.67887	0.00075	0.69290
270 cm	0.4	0.4	No	75 %	0.67727	0.00105	0.69142
300 cm	0.4	0.4	No	75 %	0.68166	0.00099	0.69578
250 cm	0.2	0.2	No	75 %	0.56011	0.00059	0.57409
270 cm	0.2	0.2	No	75 %	0.55940	0.00118	0.57361
300 cm	0.2	0.2	No	75 %	0.56053	0.00119	0.57475
250 cm	0.1	0.1	No	75 %	0.49514	0.00044	0.50908
270 cm	0.1	0.1	No	75 %	0.49439	0.00135	0.50870
300 cm	0.1	0.1	No	75 %	0.49446	0.00122	0.50870

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6.4.3 Criticality Results for Canistered Yankee Class Fuel

This section establishes the most reactive Yankee Class fuel and the most reactive configuration of the fuel within the canister basket. These results are used to calculate the effective neutron multiplication factor for the transfer cask and storage cask assuming full moderation. Sections 6.4.3.2 through 6.4.3.4 contain the results for the basket in the transport configuration without enlarged fuel tubes, while Section 6.4.3.5 extends the evaluation results to the basket with four enlarged fuel tubes.

6.4.3.1 Most Reactive Assembly

A simplified KENO-Va calculation of the Yankee Class design basis assemblies, described in Table 6.2-2, is performed to determine the most reactive assembly. In this simplified model, a unit cell of the NAC-STC canister basket, with the stainless steel and aluminum webbing properly spaced axially, is described. Reflecting boundary conditions are imposed on the sides, top and bottom simulating an infinite array of basket cells. Using the basket cell model, a k_{eff} value was obtained for each assembly type. The results of the evaluation are shown in Table 6.4.3-3.

The results of Table 6.4.3-3 show that either the United Nuclear Type A or Type B assembly has the highest multiplication factor of the Yankee Class fuel vendor categories. As shown in Table 6.4.3-3, even though the Type A assembly has an additional fuel rod, it is difficult to resolve the difference between Type A and Type B fuel assemblies. With the additional fuel rod, the United Nuclear Type A has the highest UO_2 mass; therefore, this assembly is selected as the most reactive design basis fuel assembly and is used in subsequent cask criticality analysis.

The basket cell model described above is applied to determine the most reactive Reconfigured Fuel Assembly configuration. Based on the rod parameters for the Yankee type reconfigured assembly in Table 6.2-3, only two unique types of fuel rods are modeled. One representing the CE, Exxon, and UNC fuel rods with Zircaloy clad, and the other representing the Westinghouse steel clad fuel rods. The CE, Exxon, and UNC fuel rod group is evaluated at a bounding enrichment of 4.0 wt % ^{235}U . To ensure a maximum reactivity calculation the reconfigured assembly is modeled once with a full load, 64 rods, and once with a half load, 32 rods. The 32 rod configuration consists of evenly distributed rods in the 64 tube lattice. The reactivity evaluation of the Reconfigured Fuel Assembly assumes water ingress into the tube to rod gap

and into the rod to fuel pellet gap. The maximum reactivity CSAS25 input and output for the Reconfigured Fuel Assembly evaluation are presented in Figure 6.7-7.

Configuration	Initial Enrichment	Number of Rods	k_{eff}	σ
Intact United Nuclear Type A Assembly	4.0 wt % ^{235}U	237	0.8974	0.0009
Reconfigured - Zircaloy Clad Fuel Rods	4.0 wt % ^{235}U	64	0.6280	0.0007
Reconfigured - Zircaloy Clad Fuel Rods	4.0 wt % ^{235}U	32	0.4458	0.0006
Reconfigured - Steel Clad Fuel Rods	4.94 wt % ^{235}U	64	0.6145	0.0006

Based on this evaluation, the reconfigured assembly composed of 64 Zircaloy clad fuel rods is the most limiting reconfigured assembly. Its reactivity is significantly lower than that of the limiting intact assembly.

An evaluation of the recaged fuel assembly, containing United Nuclear fuel rods in a CE lattice, did not result in a statistically different reactivity than the United Nuclear Type A or B assembly. Moving the higher enrichment United Nuclear rods into the CE lattice increased reactivity in the recaged assembly Type B cage to a k_{eff} of 0.8983 from a CE base assembly reactivity of 0.8939.

Additional analyses were performed to account for certain United Nuclear, Combustion Engineering, Exxon and Westinghouse Type A and Type B fuel assemblies having nominally higher maximum enrichment than that considered in the design basis. The base case unit cell models of the United Nuclear, Combustion Engineering, Exxon and Westinghouse fuel were modified to increase the enrichment of the fuel to 4.03 wt % ^{235}U , 3.93 wt % ^{235}U , 4.03 wt % ^{235}U and 4.97 wt % ^{235}U , respectively. These nominal increases resulted from variances in the method in which the enrichment tolerance was applied during fabrication.

The calculated differences in reactivity for these fuel assemblies show that the increased enrichments do not result in a statistically significant increase in reactivity for all fuel types, except the Westinghouse Type A fuel. However, the reactivity of the Westinghouse Type A fuel with increased enrichment is significantly less than the United Nuclear Type A or B fuel at the nominal enrichment. No statistically significant differences in reactivity exist between the United Nuclear Type A or Type B fuels at 4.00 or 4.03 wt % ^{235}U . Therefore, the design basis analysis, which uses United Nuclear Type A fuel at 4.00 wt % ^{235}U , is adequate to demonstrate the criticality safety.

6.4.3.2 Most Reactive Mechanical Configuration

Using the fuel/basket model with the design basis fuel assembly, an evaluation of the effect of different NAC-STC basket perturbations is made. This criticality analysis determines the most reactive basket mechanical configuration by altering the nominal fuel/basket model with the design basis assembly and comparing the perturbed k_{eff} to the nominal result. If Δk_{eff} ($k_{\text{perturbed}} - k_{\text{nominal}}$) is positive, the tolerance causes an increase in reactivity. Conversely, if Δk_{eff} is negative, the tolerance causes a decrease in reactivity. Two sets of perturbations are assessed in this evaluation of the criticality control: fabrication tolerances and component movement within the basket.

Four major fabrication tolerances are evaluated: the fuel tube opening, the disk opening, the disk thickness and the disk opening placement. Modifications to the nominal fuel/basket model dimensions are made based on the basket and fuel tube tolerances. The tolerances applied in this evaluation are ± 0.0762 cm for the tube opening, ± 0.0508 cm for the disk thickness, and ± 0.0381 cm on the disk fuel tube opening size. The disk opening location tolerance is within a 0.0381 cm radius circle from the nominal position. The tolerance analysis results are:

Analysis	k_{eff}	σ	Δk_{eff}
Nominal	0.8981	0.0007	-
Fuel Tube Maximum Opening	0.9018	0.0007	0.0037
Fuel Tube Minimum Opening	0.8916	0.0007	-0.0065
Disk Maximum Opening	0.8972	0.0007	-0.0009
Disk Minimum Opening	0.8991	0.0008	0.0010
Disk Maximum Thickness	0.8987	0.0008	0.0006
Disk Minimum Thickness	0.8972	0.0008	-0.0009
Loose Packed Disk Opening	0.8974	0.0008	-0.0007
Close Packed Disk Opening	0.8993	0.0007	0.0012

The results show that the most reactive set of basket tolerances are maximum fuel tube opening, minimum disk opening, maximum disk thickness, and minimum (close packed) disk opening placement.

Increasing the fuel tube opening brings more moderator into the gap between the assembly and the tube lowering the efficiency of the neutron absorber sheets, hence increasing the reactivity of the system. Minimizing the disk opening and maximizing the disk thickness removes water from the flux trap, consequently increasing k_{eff} . Finally, decreasing the web thickness, decreases the flux trap size and also moves assemblies closer together producing an increase in k_{eff} . With respect to fabrication tolerances, this is the most reactive configuration.

Two major component movements within the basket are evaluated: the assembly within the tube and the tube within the basket. Unique to this package is the Yankee Class diagonally symmetric fuel assembly. Consequently, movement toward three corners must be evaluated as opposed to one corner for a fully symmetric assembly. This assembly produces five movement perturbations: fuel tube movement to the upper right corner, the upper left corner, the lower left corner and side to side. Shown below are the assembly movement analysis results.

Assembly Movement	Boundary Conditions	k_{eff}	σ	Δk_{eff}
Nominal	Reflective	0.8981	0.0007	-
Upper Right Corner	Mirrored	0.8954	0.0007	-0.0027
Upper Right Corner	Periodic	0.8943	0.0007	-0.0038
Lower Left Corner	Mirrored	0.8977	0.0007	-0.0004
Lower Left Corner	Periodic	0.8978	0.0008	-0.0003
Upper Left Corner	Mirrored	0.8963	0.0007	-0.0018
Upper Left Corner	Periodic	0.8961	0.0008	-0.0020
Right Side	Mirrored	0.8949	0.0007	-0.0032
Right Side	Periodic	0.8951	0.0007	-0.0030
Left Side	Mirrored	0.8978	0.0007	-0.0003
Left Side	Periodic	0.8972	0.0007	-0.0009

These results show that the most reactive assembly position is centered within the basket tube.

Similar to the fuel assembly movement analysis, five possible fuel tube movements are evaluated: the upper right corner, the upper left corner, the lower left corner and side to side. Mirror and periodic boundary conditions on the sides of the model are evaluated.

The results of the tube movement evaluations are:

Tube Movement	Boundary Conditions	k_{eff}	σ	Δk_{eff}
Nominal	Reflective	0.8981	0.0007	-
Upper Right Corner	Mirrored	0.8999	0.0007	0.0018
Upper Right Corner	Periodic	0.8979	0.0007	-0.0002
Lower Left Corner	Mirrored	0.8984	0.0008	0.0003
Lower Left Corner	Periodic	0.8962	0.0007	-0.0019
Upper Left Corner	Mirrored	0.8991	0.0008	0.0010
Upper Left Corner	Periodic	0.8959	0.0007	-0.0022
Right Side	Mirrored	0.9005	0.0008	0.0024
Right Side	Periodic	0.8966	0.0007	-0.0015
Left Side	Mirrored	0.8968	0.0007	-0.0013
Left Side	Periodic	0.8976	0.0007	-0.0005

These results indicate that the most reactive fuel tube location is shifted to the right side of the tube with mirrored boundary conditions. This result is reasonable given the orientation of the assembly. Shifting the tube to the right side with mirrored boundary conditions moves a complete fuel rod row of two assemblies closer together, hence, pushing the largest amount of fuel together and minimizing the flux trap gap between tubes. In general, these results show that moving the tubes towards each other with the fuel assembly centered in the tube is the most reactive component configuration.

Based on the canistered fuel/basket model, the most reactive mechanical configuration occurs with the assemblies centered in the tubes, fuel tubes moved toward the center of the basket, maximum fuel tube opening, minimum disk opening, maximum disk thickness and close packed disk opening locations. The most reactive configuration documented by the fuel/basket analysis serves as the base model for the normal and accident analyses optimum moderation studies.

Directly loaded basket analyses indicate that the assembly centered in tube configuration may not represent the most reactive configuration in the cask analysis. The fuel/basket model clusters the fuel in groups of four (mirrored boundary), or shifts the fuel to one side of the tube (periodic boundary) and therefore does not represent the closest fuel material approach feasible in a radial inward moved model. To document the maximum reactivity configuration both tube and assembly movement analysis are repeated in the full cask model.

The k_{eff} of these analysis are compared to the nominal cask model:

Position	k_{eff}	σ	Δk_{eff}
Nominal	0.8637	0.0007	---
Tubes Moved Toward the Basket Center	0.8689	0.0008	0.0052
Tubes Moved Toward the Basket Shell	0.8596	0.0008	-0.0041
Assemblies Moved Toward the Basket Center	0.8677	0.0007	0.0040
Assemblies Moved Toward the Basket Shell	0.8590	0.0008	-0.0047

Based on the cask analysis of the basket model without enlarged fuel tubes, moving the assembly toward the cask center configuration adds a Δk_{eff} of 0.004 to the reactivity of the nominal configuration. The model documented as the worst-case mechanical configuration in the fuel/basket and enlarged fuel tube evaluations is not adjusted from its assembly-centered configuration. The Δk_{eff} associated with the assembly movement is accounted for by adding the Δk_{eff} of 0.004 to the KENO-Va neutron multiplication factor (k_{eff}) during k_s calculations.

6.4.3.3 Normal Conditions

Yankee Class fuel assemblies will be sealed inside a canister that is welded shut. Consequently, the canistered fuel is dry under normal conditions of loading and transport. Criticality results under normal conditions exclude variations in moderator density, but include cask center-to-center spacing from 250 cm (touching) to 300 cm. Moderator density is taken to be 0.0001 g/cc (dry). The results for normal conditions of transport are shown in Table 6.4.3-1. Table 6.4.3-1 shows that cask reactivity is relatively insensitive to variations in cask center-to-center spacing. This results in a k_{eff} of 0.4580 ± 0.0006 . The CSAS25 input and output for this case is shown in Figures 6.7-3 and 6.7-4, respectively. For conservatism a cask criticality analysis of a flooded, nominal condition, cask array is performed. The maximum reactivity for this configuration is a k_{eff} of 0.8761 ± 0.0007 .

Including statistical and method uncertainties, all results for the normal condition are below the 0.95 NRC criticality safety limit. Thus, compliance with 10 CFR 71.55 (b) and (d) as well as 10 CFR 71.75 (a) is demonstrated.

6.4.3.4 Hypothetical Accident Conditions

Criticality results under hypothetical accident conditions include variations in moderator density from 1.0 g/cc to 0.1 g/cc (dry) as well as cask center-to-center spacing from 250.698 cm (touching) to 300 cm. The results are shown in Table 6.4.3-2. Under accident conditions, the

cask and fuel is considered to be fully moderated as described in Section 6.1.2. The maximum k_{eff} , including uncertainties, for this situation is 0.9014. The CSAS25 input and output for this case is shown in Figures 6.7-5 and 6.7-6, respectively.

Including statistical and method uncertainties, all results for the accident condition are well below the 0.95 NRC criticality safety limit. Thus, compliance with 10 CFR 71.55 (e) is demonstrated.

6.4.3.5 Hypothetical Accident Evaluation for a Basket Containing Enlarged Fuel Tubes

The maximum reactivity, fully moderated, cask model is evaluated with four enlarged fuel tubes replacing the standard (neutron absorber sheets on four sides) fuel tube on the basket periphery. As expected, the reactivity of these systems increases slightly due the increased neutron interaction between fuel tubes in those locations where neutron absorber sheets were removed. Adjusting for the 0.004 Δk_{eff} associated with the assembly movement in the tubes, results in a maximum bias and uncertainty adjusted k_{eff} (k_s) of 0.9183 for the hypothetical accident condition involving full moderator intrusion. Figure 6.7-8 shows input and output files for the enlarged fuel tube case. Transport maximum reactivities for the enlarged fuel tube basket are, therefore, well below the 0.95 criticality safety limit.

6.4.3.6 Evaluation of Non-solid Replacement Rods in Yankee Class Fuel

The Yankee Class spent fuel inventory contains a limited number of fuel assemblies with replacement rods. These replacement rods include displacement rods and fill rods. Any assembly that contains fill rods that displace at least the same amount of moderator (in the fully flooded condition, including the fuel-clad gap) as the nominal fuel rods is allowed to be loaded into the NAC-STC. This is permissible because the modified assembly would be less reactive than its original configuration due to a lower fissile mass.

Some Yankee Class fuel assemblies have had up to 9 fuel rods replaced with fill rods that will not displace the same amount of moderator (in the fully flooded condition, including the fuel-clad gap) as the original fuel rods. The subject assemblies have hollow Zircaloy fill rods that contain stainless steel slugs or Zircaloy slugs. Fully flooded unit cell analyses are performed to evaluate these assemblies. The first unit cell analysis models the most reactive United Nuclear Type A assembly with no fill rods and floods the fuel-clad gap of all fuel rods to establish a base case reactivity. The second set of analyses models two geometries of a United Nuclear Type A assembly with 12 fill rods that contain stainless steel slugs. The gap between the cladding and

the slug or pellets in all rods is flooded. The diameter of the stainless steel slugs ranges from 0.265 to 0.275 inch. However, stainless steel slug diameters of up to 0.308 inches are evaluated. The results of the analyses, documented in Table 6.4.3-4, show that replacing up to 12 fuel rods with fill rods that contain stainless steel slugs with a minimum diameter of 0.265 inches decreases the system reactivity significantly.

The third set of analyses models multiple geometries of a United Nuclear Type A assembly with 12 fill rods that contain Zircaloy slugs. These models also flood the gap between the cladding and the slug or pellets of all rods. The diameter of the Zircaloy slugs ranges from 0.290 to 0.308 inch. The change in reactivity between these models and the base case, documented in Table 6.4.3-5, demonstrates that replacing up to 12 fuel rods with fill rods that contain Zircaloy slugs with a minimum diameter of 0.290 inches does not have a statistically significant impact on system reactivity. Therefore, Yankee Class fuel assemblies having up to 12 fuel rods replaced with non-solid fill rods that contain stainless steel slugs with a minimum diameter of 0.265 inches, or Zircaloy slugs with a minimum diameter of 0.290 inches, are allowed contents.

Six United Nuclear Type B assemblies in the Yankee Class inventory have two displacement rods that vary in geometry from the displacement rod dimensions listed in Table 6.2-1. To analyze these assemblies, the unit cell model of the United Nuclear Type B assembly referred to in Section 6.4.3.1 is modified to model the subject displacement rods as empty rod positions. The change in reactivity due to this configuration, $\Delta k_{\text{eff}} = 0.00021$, is within the statistical uncertainty of the code. Based on this evaluation, the United Nuclear Type B assemblies displacement rod dimension has no noticeable impact on system reactivity, and any displacement rod dimension is permissible for loading.

6.4.3.7 Basket Containing Damaged Fuel Cans

To provide flexibility and to allow loading of damaged fuel rods without placement of the damaged rods into the Yankee Class Reconfigured Fuel Assembly, a damaged fuel can basket configuration is evaluated. To accommodate damaged fuel assemblies, the four fuel tubes have been replaced with screened damaged fuel cans. The damaged fuel can is designed to preclude the release of pellets and gross particulate into the canister cavity. The evaluation of a canister loaded with four (4) damaged fuel cans considers each damaged fuel can to contain a Yankee Class assembly with up to 20 damaged or missing fuel rods and considers 100% dispersal of the fuel from these rods within the damaged fuel can.

All of the spent fuel assemblies containing fuel rods that are classified as damaged shall be stored in a damaged fuel can. The damaged fuel cans are located in the four corner locations of the basket, as shown in Figure 6.3-5.

The analysis of a canister containing four damaged fuel cans is performed in multiple steps. Damaged fuel is evaluated in a missing rod configuration followed by a loose fuel evaluation. The loose fuel evaluation considers the fissile material to be located within the fuel rod lattice or above and below the active fuel region, but within the fuel can. The evaluation is performed within the transport cask overpack.

Missing Rod Geometry

Initially, the most reactive missing rod geometry is determined for each Yankee Class fuel type. Assuming that a damaged fuel rod is a missing or empty fuel rod location is conservative as this increases the H/U ratio of the undermoderated assembly. Evaluating up to 20 missing rods in various configurations establishes an increased H/U ratio within the assembly and thus the most reactive (20) missing rod or damaged rod geometry. The determination of the most reactive missing rod geometry is conducted using the fuel tube unit cell models employed in Section 6.4.3.1. The results of this evaluation, which are documented in Tables 6.4.3-6 through 6.4.3-9, show that the most reactive missing rod geometry occurs with United Nuclear Type A fuel assemblies that have all 20 damaged rods modeled as missing rods. This missing rod geometry is shown in Figure 6.4.3-1.

The most reactive missing rod geometry is then modeled in the damaged fuel can in each of the four corner locations of the basket to provide an initial accident condition reactivity. In the mechanical configuration documented in Section 6.4.3.2, this loading configuration increases the reactivity of the system, $\Delta k_{\text{eff}} = 0.00435$, over that containing intact fuel in the enlarged fuel tubes. Therefore, a detailed study is performed to determine the reactivity of the system resulting from maximizing interaction between the intact assemblies in the basket and the contents of the damaged fuel cans. Maximum interaction occurs when the orientation of each assembly is such that the two sides of the intact assemblies that have a complete row of fuel rods face the two complete fuel rod rows of the fuel assemblies in damaged fuel cans. In addition, fuel assemblies and fuel tubes are shifted toward the damaged fuel can location. This configuration, damaged fuel cans containing assemblies with missing rods combined with a basket geometry that maximizes interaction between all contents of the basket, increases reactivity, Δk_{eff} , by 0.02461. This loading configuration, which results in a system reactivity ($k_{\text{eff}} \pm \sigma$) of $0.92480 \pm$

0.00073, is shown in Figure 6.4.3-2. The input/output file for this case is provided in Figure 6.7-11.

Dispersed Fuel Evaluation

To determine the effect on reactivity due to the dispersed damaged fuel occupying the void regions within each damaged fuel can, all dispersed fuel in each analysis is modeled as a homogeneous mixture of UO_2 and water. The volume fractions of the fuel versus the water are varied from 0%-100%. By varying the fuel fraction up to 100%, this evaluation addresses UO_2 masses significantly larger than what is available from 20 damaged Yankee Class fuel rods.

In the first dispersed fuel evaluation, the loose fuel is evaluated between the remaining rods of the most reactive missing rod geometry. The results of this analysis, provided in Table 6.4.3-10, show that this scenario decreases the reactivity of the system compared to just placing the most reactive missing rod array within each can. This is because adding fuel to the bounding missing rod array, with an increased moderator to fuel ratio, reduces the reactivity of the system as this effectively returns the system to an undermoderated state.

Loose fuel is also considered above and below the active fuel region of the most reactive missing rod array. Locating fuel outside the fuel rod active fuel region may also place fissile material outside the neutron absorber sheet axial extends. Previous criticality evaluations were based on an infinite height, repeating stack of aluminum and steel disks separated by spacers, which would not adequately account for this material. This analysis is, therefore, performed within a finite height cask model that represents the canister cavity in two regions: the region of the basket between the top and bottom support disks, and the regions within the transportable storage canister cavity that are above and below the top and bottom support disks. This model conservatively extends the active fuel height of all assemblies to match the disk elevations and conservatively truncates neutron absorber sheet elevation to the same height. A mirror boundary is placed in the middle of the region between the ends of the truncated neutron absorber and the extent of the canister cavity. This conservatively models an infinite height, alternating stacks of active fuel with neutron absorber sheet coverage and the regions without coverage. For simplicity, the height of the region below the neutron poison is modeled with the same height as the region above the neutron poison. Figure 6.4.3-3 provides a graphical representation of the various axial regions of the model.

This height is modeled first at 7.5 cm to conservatively encompass half the minimum height of the region between the neutron absorber and the canister bottom (~16.5 cm). By placing a

mirror boundary condition at the end of each 7.5 cm region, an effective 15 cm of fuel debris is uncovered. This model conservatively replaces the lids with active fuel and effectively eliminates axial leakage. The 7.5 cm height is then increased to 15 cm to encompass half the maximum height of the region between the neutron absorber and the shield lid (~28 cm).

The results of these studies, provided in Tables 6.4.3-11 and 6.4.3-12, show the possible mixture of fuel and water above and below the neutron poison sheet coverage; therefore, mixtures above and below the active fuel region will not increase the reactivity of the system beyond that of the infinite height model of each damaged fuel can containing an assembly with the most reactive (20) missing rod array, $k_{\text{eff}} \pm 2\sigma = 0.92480 \pm 0.00073$.

Evaluation of a Preferential Flooding and Uneven Draining

Preferential flooding of the canister cavity and the damaged fuel can cavity is also considered. The model utilized—an infinite height, repeating stack of aluminum and steel disks separated by spacers—considers the most reactive fuel assemblies and the most reactive damaged fuel can contents. The density of the moderator filling the canister cavity is initially set to zero to model a dry canister cavity, while the void regions in each damaged fuel can are filled with full density moderator. The density of the moderator filling the canister cavity is then incrementally increased to full density. A study is also performed with a fully flooded canister cavity, while the density of the moderator occupying the void regions in each damaged fuel can is initially set to zero and then incrementally increased to full density. The results of these moderator density variations, provided in Table 6.4.3-13, demonstrate that the most reactive configuration for both the canister cavity and the damaged fuel cans occurs with full density water.

A case modeling a partially flooded canister loaded with four fully flooded damaged fuel cans in an uneven draining condition (post-accident) is also evaluated. The geometry of this case is taken from the axially finite basket model with 7.5 cm regions above and below the active fuel and, thus, above and below the neutron absorber. The canister cavity is modeled with full density moderator along the entire axial extent of the active fuel. The four damaged fuel cans are hypothetically assumed to remain fully flooded along the entire axial extent of each can. The material occupying the 7.5 cm regions above and below the active fuel in each damaged fuel can is taken to be 10% UO_2 and 90% water. As shown in Table 6.4.3-11, this damaged fuel can content results in the maximum increase in reactivity compared to the same geometry with no loose damaged fuel. The UO_2 volume fraction of the loose damaged fuel mixture is closer to what would be expected from only 20 rods than the UO_2 mixture volume fractions employed in

the other cases. The resulting geometry, shown in Figure 6.4.3-4, bounds a hypothetical uneven draining of a canister containing damaged fuel cans. The calculated reactivity of this configuration, $k_{\text{eff}} \pm 2\sigma = 0.92004$, is within the statistical uncertainty band of the fully flooded condition of the same geometry.

Summary of Most Reactive Fuel Loading for a Basket Containing Damaged Fuel Cans

All the criticality safety analyses result in a system reactivity that is bounded by the infinite height model of a fully flooded canister loaded with 32 intact United Nuclear Type A assemblies and four damaged fuel cans, each containing a United Nuclear Type A assembly with a maximum of 20 damaged or missing fuel rods. This configuration results in a system reactivity ($k_{\text{eff}} \pm \sigma_{\text{mc}}$) of 0.92480 ± 0.00073 . The maximum system reactivity for this configuration, $k_{\text{eff}} + 2\sigma_{\text{mc}} = 0.92625$, is less than the USL of 0.9361. The maximum system reactivity including the code bias and the bias uncertainty, $k_s = k_{\text{eff}} + 0.0052 + (0.0087^2 + (2\sigma_{\text{mc}})^2)^{(1/2)}$, is calculated to be 0.93883, which is below 0.95. Thus, assemblies with up to 20 damaged or missing rods are allowed contents in a damaged fuel can.

Figure 6.4.3-1 Most Reactive Missing Rod Geometry

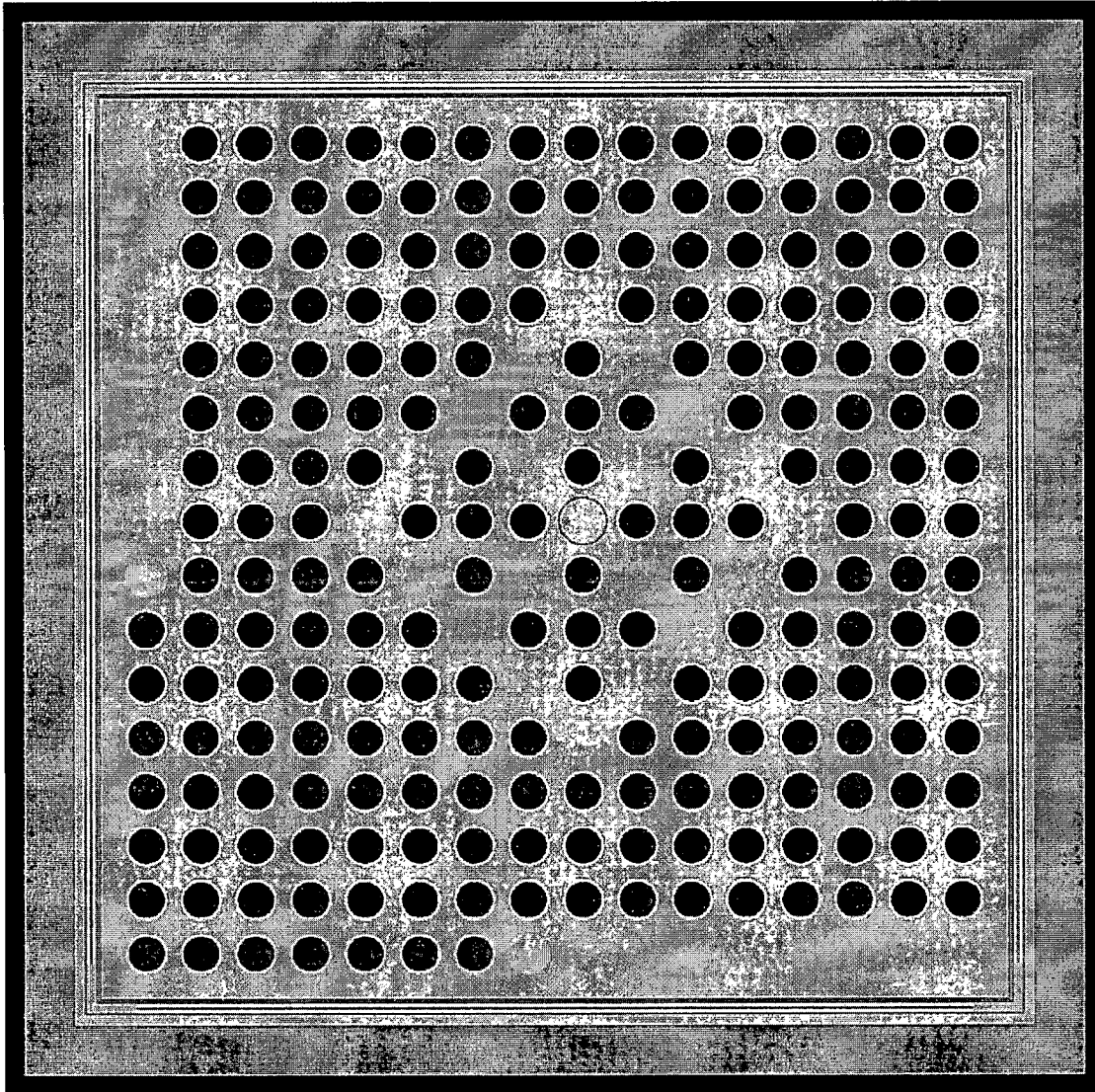


Figure 6.4.3-2 Most Reactive Missing Rod Array in the Damaged Fuel Can

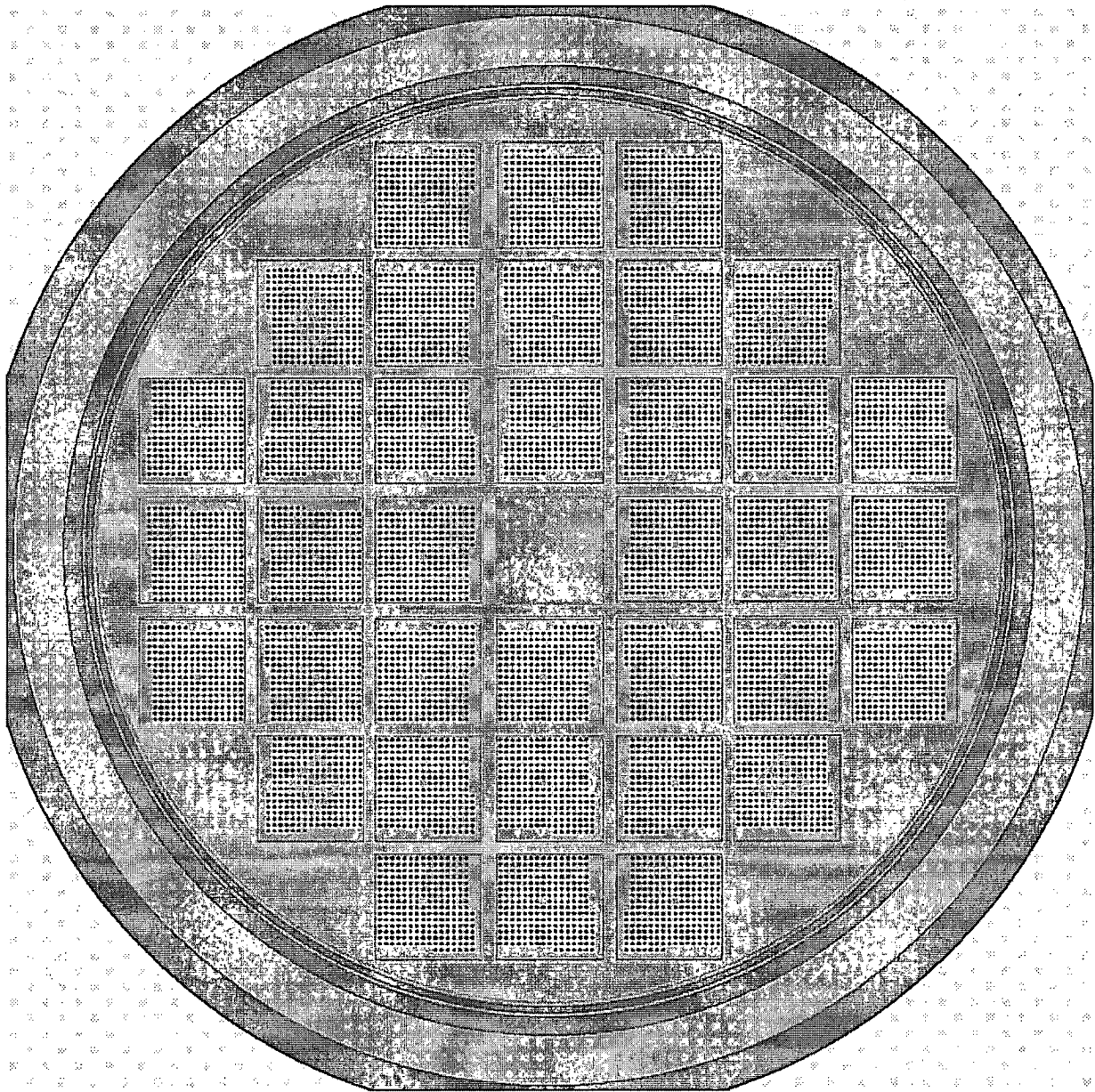


Figure 6.4.3-3 Damaged Fuel Can – 7.5 cm Regions without Neutron Absorber Coverage

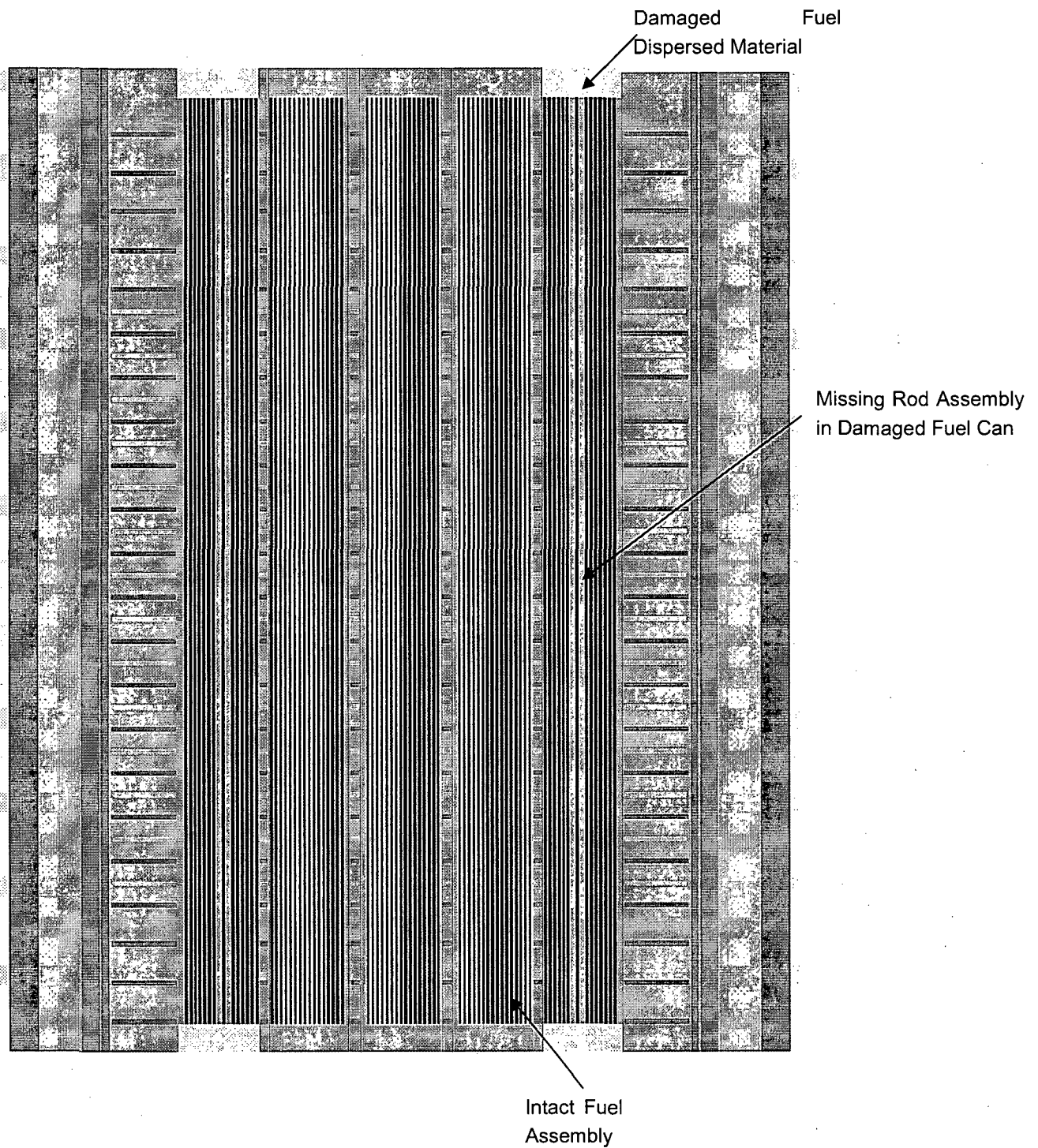


Figure 6.4.3-4 Hypothetical Uneven Drain Down Model

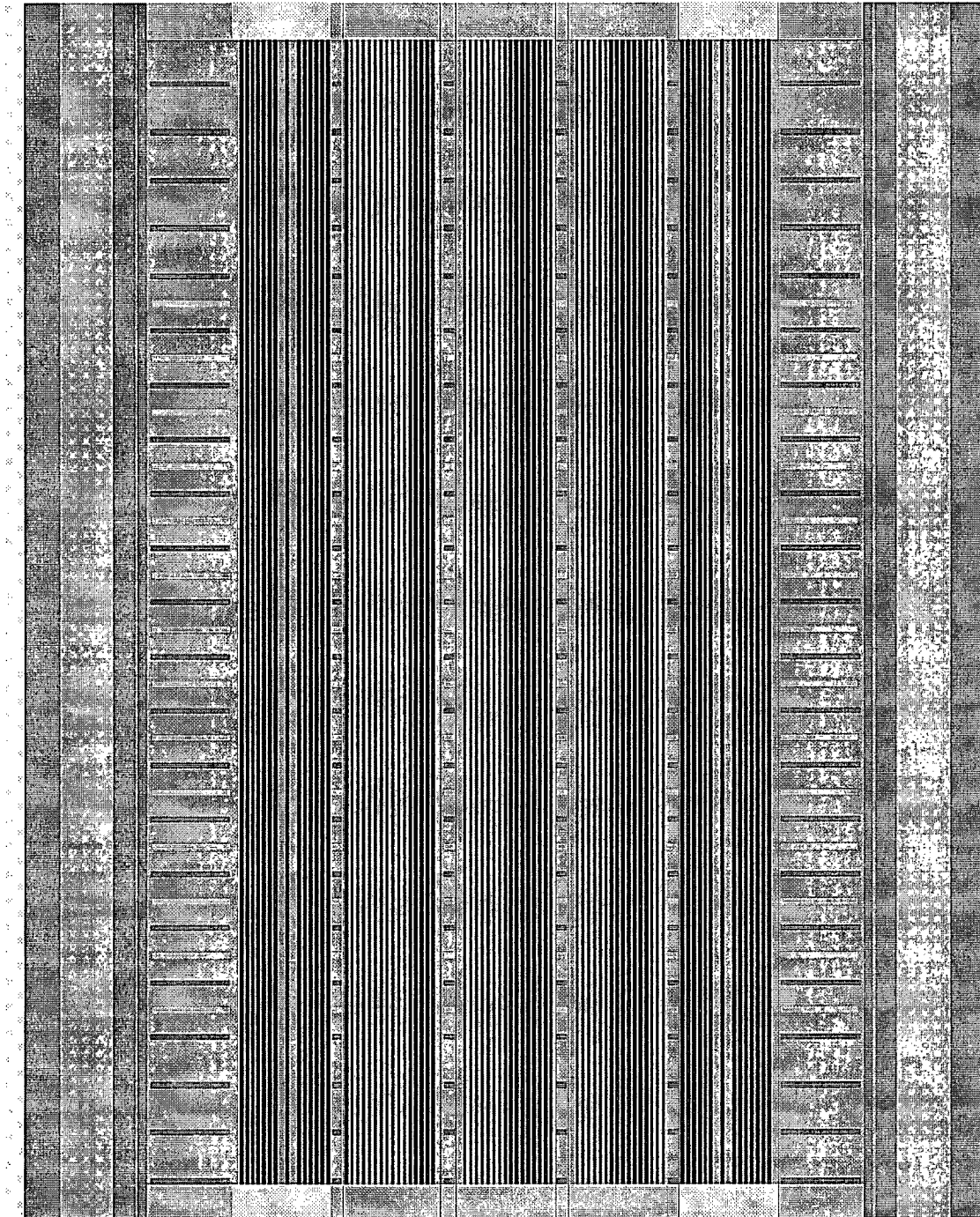


Table 6.4.3-1 Criticality Results for Normal Conditions of Transport of Yankee-MPC Canistered Fuel

Cask Pitch (cm)	H ₂ O Interior	H ₂ O Exterior	Neutron Shield	¹⁰ B	k _{eff}	σ	k _s
250.698	0.0001	1.0	Yes	75.0%	0.4573	0.00067	0.4753
270	0.0001	1.0	Yes	75.0%	0.4557	0.00062	0.4737
300	0.0001	1.0	Yes	75.0%	0.4566	0.00066	0.4746
250.698	0.0001	0.8	Yes	75.0%	0.4569	0.00068	0.4749
270	0.0001	0.8	Yes	75.0%	0.4564	0.00071	0.4744
300	0.0001	0.8	Yes	75.0%	0.4566	0.00064	0.4746
250.698	0.0001	0.6	Yes	75.0%	0.4580	0.00064	0.4760
270	0.0001	0.6	Yes	75.0%	0.4570	0.00069	0.4750
300	0.0001	0.6	Yes	75.0%	0.4579	0.00066	0.4759
250.698	0.0001	0.4	Yes	75.0%	0.4573	0.00046	0.4752
270	0.0001	0.4	Yes	75.0%	0.4575	0.00049	0.4755
300	0.0001	0.4	Yes	75.0%	0.4566	0.0006	0.4746
250.698	0.0001	0.2	Yes	75.0%	0.4557	0.00063	0.4737
270	0.0001	0.2	Yes	75.0%	0.4577	0.00073	0.4758
300	0.0001	0.2	Yes	75.0%	0.4576	0.00067	0.4756
250.698	0.0001	0.1	Yes	75.0%	0.4562	0.00061	0.4742
270	0.0001	0.1	Yes	75.0%	0.4566	0.00068	0.4746
300	0.0001	0.1	Yes	75.0%	0.4572	0.00067	0.4752

Table 6.4.3-2 Criticality Results for Yankee-MPC Canistered Fuel Hypothetical Accident Conditions

Cask Pitch (cm)	H ₂ O Interior	H ₂ O Exterior	Neutron Shield	¹⁰ B	k _{eff}	σ	k _s
250.698	1.0	1.0	No	75.0%	0.8830	0.00074	0.9011
270	1.0	1.0	No	75.0%	0.8819	0.00075	0.8999
300	1.0	1.0	No	75.0%	0.8834	0.00075	0.9014
250.698	0.8	0.8	No	75.0%	0.8328	0.0008	0.8509
270	0.8	0.8	No	75.0%	0.8334	0.00083	0.8514
300	0.8	0.8	No	75.0%	0.8312	0.00084	0.8493
250.698	0.6	0.6	No	75.0%	0.7725	0.00089	0.7906
270	0.6	0.6	No	75.0%	0.7728	0.00089	0.7908
300	0.6	0.6	No	75.0%	0.7742	0.0009	0.7923
250.698	0.4	0.4	No	75.0%	0.6994	0.00094	0.7175
270	0.4	0.4	No	75.0%	0.6997	0.00107	0.7178
300	0.4	0.4	No	75.0%	0.6973	0.00103	0.7154
250.698	0.2	0.2	No	75.0%	0.6086	0.00121	0.6268
270	0.2	0.2	No	75.0%	0.6113	0.00113	0.6295
300	0.2	0.2	No	75.0%	0.6098	0.0005	0.6278
250.698	0.1	0.1	No	75.0%	0.5585	0.0009	0.5766
270	0.1	0.1	No	75.0%	0.5578	0.00097	0.5759
300	0.1	0.1	No	75.0%	0.5580	0.00094	0.5761

Table 6.4.3-3 Assembly Type Reactivity Evaluations

Assembly	Initial Enrichment	k_{eff}	σ
Westinghouse Type A ¹	4.94 wt% ²³⁵ U	0.8642	0.00105
Westinghouse Type B	4.94 wt% ²³⁵ U	0.8664	0.00102
United Nuclear Type A	4.00 wt% ²³⁵ U	0.8974	0.00087
United Nuclear Type B ²	4.00 wt% ²³⁵ U	0.8974	0.00106
Exxon - ANF Type A	4.00 wt% ²³⁵ U	0.8870	0.00111
Exxon - ANF Type B	4.00 wt% ²³⁵ U	0.8877	0.00111
Combustion Engineering Type A	3.90 wt% ²³⁵ U	0.8943	0.00060
Combustion Engineering Type B	3.90 wt% ²³⁵ U	0.8939	0.00163

1. At an enrichment of 4.97 wt % ²³⁵U, k_{eff} is 0.8670.
2. At an enrichment of 4.03 wt % ²³⁵U, k_{eff} is 0.8992.

Table 6.4.3-4 Most Reactive Fuel Containing Non-solid Fill Rods with Stainless Steel Slugs

Patter n	Slug Diam (in)	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
-----	-----	0.90565	0.00074	-----	-----
a	0.265	0.89351	0.00073	-0.01214	-16.4
a	0.270	0.89382	0.00072	-0.01183	-16.0
a	0.275	0.89284	0.00073	-0.01281	-17.3
a	0.280	0.89099	0.00073	-0.01466	-19.8
a	0.285	0.89071	0.00073	-0.01494	-20.2
a	0.290	0.89009	0.00074	-0.01556	-21.0
a	0.295	0.88806	0.00074	-0.01759	-23.8
a	0.300	0.88741	0.00073	-0.01824	-24.6
a	0.305	0.88796	0.00071	-0.01769	-23.9
a	0.308	0.88555	0.00072	-0.02010	-27.2
b	0.265	0.89278	0.00073	-0.01287	-17.4
b	0.270	0.89141	0.00073	-0.01424	-19.2
b	0.275	0.89197	0.00075	-0.01368	-18.5
b	0.280	0.88887	0.00074	-0.01678	-22.7
b	0.285	0.88824	0.00074	-0.01741	-23.5
b	0.290	0.88790	0.00072	-0.01775	-24.0
b	0.295	0.88712	0.00073	-0.01853	-25.0
b	0.300	0.88452	0.00074	-0.02113	-28.6
b	0.305	0.88358	0.00073	-0.02207	-29.8
b	0.308	0.88588	0.00072	-0.01977	-26.7

Table 6.4.3-5 Most Reactive Fuel Non-solid Fill Rods with Zircaloy Slugs

Pattern	Slug Diam (in)	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
-----	-----	0.90565	0.00074	-----	-----
a	0.290	0.90708	0.00071	0.00143	1.9
a	0.295	0.90692	0.00033	0.00127	1.7
a	0.300	0.90684	0.00074	0.00119	1.6
a	0.305	0.90576	0.00072	0.00011	0.1
a	0.308	0.90542	0.00073	-0.00023	-0.3
b	0.290	0.90662	0.00071	0.00097	1.3
b	0.295	0.90566	0.00076	0.00001	0.0
b	0.300	0.90684	0.00025	0.00119	1.6
b	0.305	0.90524	0.00073	-0.00041	-0.6
b	0.308	0.90455	0.00074	-0.00110	-1.5
c	0.290	0.90646	0.00073	0.00081	1.1
c	0.295	0.90528	0.00033	-0.00037	-0.5
c	0.300	0.90456	0.00077	-0.00109	-1.5
c	0.305	0.90657	0.00072	0.00092	1.2
c	0.308	0.90539	0.00073	-0.00026	-0.4
d	0.290	0.90562	0.00072	-0.00003	0.0
d	0.295	0.90685	0.00033	0.00120	1.6
d	0.300	0.90606	0.00076	0.00041	0.6
d	0.305	0.90493	0.00077	-0.00072	-1.0
d	0.308	0.90522	0.00074	-0.00043	-0.6
e	0.290	0.90711	0.00018	0.00146	1.97
e	0.295	0.90648	0.00033	0.00083	1.1
e	0.300	0.90606	0.00076	0.00041	0.6
e	0.305	0.90493	0.00077	-0.00072	-1.0
e	0.308	0.90522	0.00074	-0.00043	-0.6

Table 6.4.3-6 United Nuclear Type Removed Rod Results

# of Removed Rods	Type A				Type B			
	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
01	0.90152	0.00123	-0.02747	-23.5	0.90235	0.00121	-0.02593	-20.9
01	0.90069	0.00106	-0.02830	-24.2	0.89816	0.00114	-0.03012	-24.3
02	0.89986	0.00121	-0.02913	-24.9	0.90229	0.00108	-0.02599	-21.0
02	0.89967	0.00115	-0.02932	-25.1	0.90113	0.00113	-0.02715	-21.9
04	0.90466	0.00123	-0.02433	-20.8	0.90507	0.00111	-0.02321	-18.7
04	0.90459	0.00113	-0.02440	-20.9	0.90569	0.00115	-0.02259	-18.2
08	0.91344	0.00119	-0.01555	-13.3	0.91027	0.00118	-0.01801	-14.5
08	0.91342	0.00121	-0.01557	-13.3	0.91210	0.00113	-0.01618	-13.0
12	0.91483	0.00108	-0.01416	-12.1	0.91777	0.00121	-0.01051	-8.5
12	0.91930	0.00117	-0.00969	-8.3	0.91751	0.00125	-0.01077	-8.7
16	0.92239	0.00116	-0.00660	-5.6	0.92272	0.00120	-0.00556	-4.5
16	0.91899	0.00113	-0.01000	-8.5	0.91744	0.00115	-0.01084	-8.7
20	0.92612	0.00111	-0.00287	-2.5	0.92800	0.00118	-0.00028	-0.2
20	0.92833	0.00116	-0.00066	-0.6	0.92603	0.00119	-0.00225	-1.8
20	0.92899	0.00117	-----	-----	0.92626	0.00115	-0.00202	-1.6
20	0.92666	0.00109	-0.00233	-2.0	0.92828	0.00124	-----	-----

Table 6.4.3-7 Combustion Engineering Type Removed Rod Results

# of Removed Rods	Type A				Type B			
	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
01	0.89413	0.00114	-0.02906	-24.6	0.89397	0.00114	-0.02997	-25.8
01	0.89648	0.00109	-0.02671	-22.6	0.89481	0.00118	-0.02913	-25.1
02	0.89594	0.00115	-0.02725	-23.1	0.89584	0.00119	-0.02810	-24.2
02	0.89507	0.00117	-0.02812	-23.8	0.89697	0.00122	-0.02697	-23.2
04	0.90229	0.00109	-0.02090	-17.7	0.90213	0.00120	-0.02181	-18.8
04	0.90108	0.00113	-0.02211	-18.7	0.90048	0.00115	-0.02346	-20.2
08	0.90533	0.00116	-0.01786	-15.1	0.90822	0.00114	-0.01572	-13.6
08	0.90612	0.00108	-0.01707	-14.5	0.90661	0.00121	-0.01733	-14.9
12	0.90941	0.00114	-0.01378	-11.7	0.90993	0.00113	-0.01401	-12.1
12	0.91232	0.00118	-0.01087	-9.2	0.91239	0.00122	-0.01155	-10.0
16	0.91807	0.00114	-0.00512	-4.3	0.9178	0.00116	-0.00614	-5.3
16	0.91282	0.00115	-0.01037	-8.8	0.91421	0.00112	-0.00973	-8.4
20	0.92263	0.00117	-0.00056	-0.5	0.92254	0.00114	-0.00140	-1.2
20	0.92042	0.00114	-0.00277	-2.3	0.91982	0.00119	-0.00412	-3.6
20	0.92212	0.0012	-0.00107	-0.9	0.92394	0.00116	-----	-----
20	0.92319	0.00118	-----	-----	0.9221	0.00116	-0.00184	-1.6

Table 6.4.3-8 Exxon Type Removed Rod Results

# of Removed Rods	Type A				Type B			
	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
01	0.89171	0.00118	-0.02705	-23.9	0.89017	0.00113	-0.02604	-24.8
01	0.88843	0.00119	-0.03033	-26.8	0.89000	0.00112	-0.02621	-25.0
02	0.88992	0.00107	-0.02884	-25.5	0.89067	0.00109	-0.02554	-24.3
02	0.89274	0.00108	-0.02602	-23.0	0.88884	0.00111	-0.02737	-26.1
04	0.89287	0.00115	-0.02589	-22.9	0.89334	0.00119	-0.02287	-21.8
04	0.89416	0.00112	-0.02460	-21.8	0.89293	0.00114	-0.02328	-22.2
08	0.90302	0.00116	-0.01574	-13.9	0.89925	0.00115	-0.01696	-16.2
08	0.89969	0.00116	-0.01907	-16.9	0.90053	0.00110	-0.01568	-14.9
12	0.90492	0.00123	-0.01384	-12.2	0.90562	0.00115	-0.01059	-10.1
12	0.90812	0.00102	-0.01064	-9.4	0.90544	0.00120	-0.01077	-10.3
16	0.91225	0.00125	-0.00651	-5.8	0.90952	0.00121	-0.00669	-6.4
16	0.90747	0.00112	-0.01129	-10.0	0.90679	0.00121	-0.00942	-9.0
20	0.91876	0.00113	-----	-----	0.91433	0.00116	-0.00188	-1.8
20	0.91425	0.00114	-0.00451	-4.0	0.91254	0.00115	-0.00367	-3.5
20	0.91498	0.00116	-0.00378	-3.3	0.91183	0.00118	-0.00438	-4.2
20	0.91687	0.00115	-0.00189	-1.7	0.91621	0.00105	-----	-----

Table 6.4.3-9 Westinghouse Type A Removed Rod Results

# of Removed Rods	Type A				Type B			
	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$	k_{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
01	0.86696	0.00109	-0.02844	-26.8	0.86772	0.00108	-0.02809	-26.8
01	0.86502	0.00112	-0.03038	-28.7	0.86720	0.00108	-0.02861	-27.2
02	0.86736	0.00102	-0.02804	-26.5	0.86832	0.00105	-0.02749	-26.2
02	0.86715	0.00107	-0.02825	-26.7	0.87009	0.00110	-0.02572	-24.5
04	0.87038	0.00105	-0.02502	-23.6	0.87128	0.00107	-0.02453	-23.4
04	0.87110	0.00109	-0.02430	-22.9	0.87058	0.00109	-0.02523	-24.0
08	0.87648	0.00105	-0.01892	-17.8	0.87938	0.00110	-0.01643	-15.6
08	0.87746	0.00107	-0.01794	-16.9	0.87817	0.00109	-0.01764	-16.8
12	0.88039	0.00108	-0.01501	-14.2	0.88366	0.00110	-0.01215	-11.6
12	0.88512	0.00104	-0.01028	-9.7	0.88429	0.00106	-0.01152	-11.0
16	0.88957	0.00113	-0.00583	-5.5	0.88829	0.00099	-0.00752	-7.2
16	0.88917	0.00107	-0.00623	-5.9	0.88933	0.00109	-0.00648	-6.2
20	0.89458	0.00113	-0.00082	-0.8	0.89565	0.00108	-0.00016	-0.2
20	0.89429	0.00109	-0.00111	-1.0	0.89233	0.00106	-0.00348	-3.3
20	0.89540	0.00106	-----	-----	0.89581	0.00105	-----	-----
20	0.89514	0.00108	-0.00026	-0.2	0.89384	0.00109	-0.00197	-1.9

Table 6.4.3-10 Damaged Fuel Can Results of Fuel-Water Mixture between Rods

Volume Fraction of UO_2 in Water	k_{eff}	σ	Δk_{eff}	$\Delta k_{\text{eff}} / \sigma$
0.0	0.92480	0.00073	-----	-----
0.1	0.91083	0.00072	-0.01397	-19.1
0.2	0.90715	0.00074	-0.01765	-24.2
0.3	0.90313	0.00073	-0.02167	-29.7
0.4	0.89873	0.00074	-0.02607	-35.7
0.5	0.89733	0.00076	-0.02747	-37.6
0.6	0.89611	0.00073	-0.02869	-39.3
0.7	0.89504	0.00073	-0.02976	-40.8
0.8	0.89374	0.00071	-0.03106	-42.5
0.9	0.89411	0.00074	-0.03069	-42.0
1.0	0.89292	0.00074	-0.03188	-43.7

Table 6.4.3-11 Damaged Fuel Can Analysis Results of Fuel-Water Mixture Outside
Neutron Absorber Coverage – 30 cm Exposed

Height of Top & Bottom Exposed Regions (cm)	Volume Fraction of UO ₂ in Water	k _{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
30	0.0	0.91886 ¹	0.00074	-----	-----
30	0.1	0.91710	0.00074	-0.00171	-2.4
30	0.2	0.91872	0.00074	-0.00009	-0.1
30	0.3	0.91753	0.00071	-0.00128	-1.8
30	0.4	0.91921	0.00073	0.00040	0.6
30	0.5	0.91811	0.00076	-0.00070	-1.0
30	0.6	0.91826	0.00077	-0.00055	-0.8
30	0.7	0.91984	0.00074	0.00103	1.5
30	0.8	0.91807	0.00074	-0.00074	-1.0
30	0.9	0.91827	0.00074	-0.00054	-0.8
30	1.0	0.91874	0.00076	-0.00007	-0.1

Notes:

1. Lower in reactivity than the 0% volume fraction case in Table 6.4.3-10 due to axial leakage from finite basket height model.

Table 6.4.3-12 Damaged Fuel Can Analysis Results of Fuel-Water Mixture Outside
Neutron Absorber Coverage – 15 cm Exposed

Height of Top & Bottom Exposed Regions (cm)	Volume Fraction of UO ₂ in Water	k _{eff}	σ	Δk_{eff}	$\Delta k_{eff} / \sigma$
15	0.0	0.91763 ¹	0.00073	-----	-----
15	0.1	0.91966	0.00073	0.00203	2.8
15	0.2	0.91791	0.00076	0.00028	0.4
15	0.3	0.91946	0.00073	0.00183	2.5
15	0.4	0.91846	0.00076	0.00083	1.1
15	0.5	0.91828	0.00074	0.00065	0.9
15	0.6	0.91770	0.00072	0.00007	0.1
15	0.7	0.91746	0.00071	-0.00017	-0.2
15	0.8	0.91753	0.00075	-0.00010	-0.1
15	0.9	0.91948	0.00074	0.00185	2.5
15	1.0	0.91881	0.00071	0.00118	1.6

Notes:

1. Lower in reactivity than the 0% volume fraction case in Table 6.4.3-10 due to axial leakage from finite basket height model.

Table 6.4.3-13 Damaged Fuel Can Preferential Flooding Analysis

Moderator Density		k_{eff}	σ	Δk_{eff}	$\Delta k_{\text{eff}} / \sigma$
TSC	Fuel Can				
1.0	1.0	0.92480 ¹	0.00073	-----	-----
0.0	1.0	0.89988	0.00076	-0.02492	-34.1
0.1	1.0	0.85082	0.00064	-0.07398	-101.3
0.2	1.0	0.84184	0.00071	-0.08296	-113.6
0.3	1.0	0.84192	0.00068	-0.08288	-113.5
0.4	1.0	0.84910	0.00073	-0.07570	-103.7
0.5	1.0	0.85830	0.00072	-0.06650	-91.1
0.6	1.0	0.86870	0.00072	-0.05610	-76.8
0.7	1.0	0.88328	0.00073	-0.04152	-56.9
0.8	1.0	0.89573	0.00074	-0.02907	-39.8
0.9	1.0	0.90964	0.00075	-0.01516	-20.8
1.0	0.0	0.88624	0.00071	-0.03856	-52.8
1.0	0.1	0.88708	0.00076	-0.03772	-51.7
1.0	0.2	0.88879	0.00073	-0.03601	-49.3
1.0	0.3	0.89028	0.00070	-0.03452	-47.3
1.0	0.4	0.89134	0.00074	-0.03346	-45.8
1.0	0.5	0.89500	0.00076	-0.02980	-40.8
1.0	0.6	0.89672	0.00075	-0.02808	-38.5
1.0	0.7	0.90249	0.00076	-0.02231	-30.6
1.0	0.8	0.90933	0.00073	-0.01547	-21.2
1.0	0.9	0.91511	0.00074	-0.00969	-13.3

1. Base case from Table 6.4.3-10 with 0.0 UO₂ in H₂O.

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6.4.4 Criticality Results for Canistered Connecticut Yankee Fuel

The criticality evaluation of the CY-MPC is performed using the MONK8a Monte Carlo Program for Nuclear Criticality Safety Analysis. The code employs JEF 2.2 point energy neutron cross-section libraries. Calculations are performed for the normal operating conditions and accident conditions (contingencies) of transport, which include optimum moderation. The evaluation considers the canister loaded with intact fuel assemblies, assemblies with Reactor Control Cluster Assemblies or Flow Mixers, fuel debris within CY reconfigured fuel assemblies and CY damaged fuel cans containing fuel assemblies with damaged cladding. Bounding fuel assembly dimensions, conservative basket dimensions, radial and axial shifting of the fuel within the basket, partial flooding of the canister and optimum moderation are also considered.

The individual MONK8a calculations execute until the standard deviation of the final result is less than or equal to 0.0008 for each of the configurations analyzed. Upon completion of each set of analyses, a reactivity comparison is made to determine the most reactive configuration.

6.4.4.1 Connecticut Yankee Fuel Loading Optimization

The CY-MPC canister has two basket configurations. The 26-assembly basket represents a canister fully loaded with 26-design basis fuel assemblies, except the Westinghouse Vantage 5H. The 24-assembly basket represents a canister and cask loaded with 24 design basis Westinghouse Vantage 5H fuel assemblies. The 24-assembly basket may be loaded with any fuel assemblies, but is evaluated using the Vantage 5H, which is the highest reactivity fuel. The two baskets are identical, except that 2 fuel loading positions of the 26-assembly basket are blocked to create the 24-assembly basket. The blocked fuel positions cannot be loaded. The models use a reflecting boundary condition on the radial surface of the transfer cask and periodic boundary conditions on the top and bottom. These conditions bound an infinite three-dimensional array of casks.

6.4.4.2 Connecticut Yankee Fuel Criticality Results

This section establishes the most reactive CY-MPC contents and the most reactive loading for each CY-MPC basket configuration. These results are used to calculate the effective neutron multiplication factor for the transport cask assuming full moderation.

6.4.4.2.1 Most Reactive Assembly – Connecticut Yankee Fuel

Each of the Connecticut Yankee fuel assembly design groupings shown in Table 6.2-4 is evaluated in the fully flooded 26-assembly basket configuration. The basket is configured assuming centered basket components with nominal dimensions. For each design grouping, the bounding fuel assembly dimensions that serve to increase moderation within the active fuel region are considered. To maximize moderation, the fuel pellet/clad gap is assumed to be flooded, and the maximum fuel rod pitch is used in conjunction with the minimum clad outer diameter and cladding thickness. To maximize the fuel volume, the maximum active fuel length is used for each design grouping. Fuel pellet diameter studies are also performed. Maximizing the fuel pellet diameter is not necessarily conservative, since the increased fuel volume displaces moderator in the flooded pellet/clad gap. In order to standardize the comparison, each assembly is evaluated with the UO_2 at 95% theoretical density.

Table 6.4.4-1 presents the results for each of the fuel assembly design groupings at three different pellet outside diameters. The pellet diameters are labeled minimum, middle and maximum and were established based on the variations of the pellet diameters within each of the design groups. The results of the evaluation demonstrate that the reactivity of the Westinghouse Vantage 5H fuel assemblies at a maximum initial enrichment of 4.61 wt % ^{235}U in the 26-assembly basket configuration is 0.9460, which exceeds the upper subcritical limit of 0.9425 (Section 6.5.2). As a result, Zircaloy-clad fuel enriched over 3.93 wt % ^{235}U may be loaded only in the 24-assembly basket configuration. Disk openings 12 and 15 in Figure 6.3-3 are blocked in the 24-assembly basket configuration because removing the fuel assemblies from these locations minimizes neutron interaction between all fuel assemblies in the system and, thus, provides the largest reactivity penalty (or reduction) from removing only two assemblies. The k_{eff} of the 4.61 wt % Zircaloy-clad Westinghouse Vantage 5H fuel assemblies centered in the 24-assembly basket configuration is 0.9197, compared to 0.9460 in the 26-assembly basket.

For the 26-assembly basket configuration, the Zircaloy-clad fuel enriched to 3.93 wt % ^{235}U bounds all stainless steel-clad fuel and the lower enrichment Zircaloy-clad fuel. The remaining analyses are, therefore, performed with either the 24-assembly basket configuration containing 4.61 wt % ^{235}U Zircaloy-clad fuel assemblies, or the 26-assembly basket configuration containing 3.93 wt % ^{235}U Zircaloy-clad fuel assemblies. As shown in Table 6.4.4-1, no statistically significant differences in reactivity (i.e., greater than 2σ or $0.0016 \Delta k_{\text{eff}}$) result from the pellet diameter variations. Therefore, further analyses will use the largest pellet diameter, as this maximizes the fuel within the system.

The CY reconfigured fuel assembly with the most reactive fuel rods is also explicitly modeled in the 26-assembly basket configuration. As shown in Section 6.4.4.2.6, detailed studies are performed demonstrating that the CY reconfigured fuel assembly containing fuel material enriched to 4.61 wt % ^{235}U is lower in reactivity than intact fuel assemblies.

6.4.4.2.2 Most Reactive Mechanical Configuration - Connecticut Yankee Fuel

Evaluations of the effect of basket dimensional perturbations are made using the 26-assembly and 24-assembly basket configurations, along with the design basis 3.93 and 4.61 wt % ^{235}U Zircaloy-clad fuel assemblies, respectively. These criticality analyses determine the most reactive basket mechanical configurations by altering the nominal model dimensional parameters and comparing the perturbed k_{eff} to the nominal result. If Δk_{eff} ($k_{\text{perturbed}} - k_{\text{nominal}}$) is positive, the tolerance causes an increase in reactivity. Conversely, if Δk_{eff} is negative, the tolerance causes a decrease in reactivity. Two types of perturbations are assessed in these evaluations: fabrication tolerances and radial movement of components within the basket.

Fabrication Tolerances

Five major fabrication tolerances are evaluated: the fuel tube opening size (± 0.03 inch), the disk opening size (± 0.015 inch) and center location within 0.015 inch, and the neutron poison sheet length and width (± 0.06 inch). The tolerances associated with each component are independently analyzed with assemblies centered in the fuel tubes and with the fuel tubes centered in the disk openings. A 26-assembly basket configuration with the design basis Zircaloy-clad fuel enriched to 4.61 wt % ^{235}U is used in these analyses. This fuel assembly was conservatively used, since a more reactive loading will be more sensitive to variations in component dimensions than a less reactive loading. The modeled width of the BORAL sheet is decreased to 7.974 inches, from a minimum toleranced width of 8.17 inches, to account for shifting of the BORAL sheet or off-center placement on the side of the fuel tube. This dimension was calculated by determining the minimum width remaining on half the tube face due to shifting to the opposite side of the tube, then conservatively applying this distance to the entire BORAL sheet width.

Table 6.4.4-2 indicates that the most reactive set of basket tolerances are: maximum fuel tube opening and minimum neutron poison material. Increasing the fuel tube opening brings more

moderator into the gap between the assembly and the tube, lowering the efficiency of the BORAL sheets and increasing the reactivity of the system. A case evaluating the combined effect of a BORAL sheet width of 7.974 inches and a tube opening width of 8.75 inches ($8.72 + 0.03$) is documented in Table 6.4.4-2.

Mechanical Perturbations

In addition to these basket tolerances, possible mechanical perturbations of the system are evaluated. Radial component shifting is analyzed to ensure that the system remains subcritical under shifted conditions. Axial shifting of components of the CY-MPC is evaluated in Section 6.4.4.4. Tolerances on the disk opening size are applied during the mechanical perturbation evaluation since an increased disk opening would provide additional movement space. The disk opening center location variance is applied since shifting the disk opening centers will allow increased interaction for the shifted tube and/or assembly. Disk opening center tolerances are applied in the same direction as the component shift studied.

Shifting analyses are performed for the 26-assembly basket configuration loaded with design basis Zircaloy-clad fuel enriched to 3.93 wt % ^{235}U , and for the 24-assembly basket configuration loaded with design basis Zircaloy-clad fuel enriched to 4.61 wt % ^{235}U . The results of these analyses are provided in Table 6.4.4-3 and Table 6.4.4-4 with the most reactive shifting patterns listed following the base case. The shifting patterns considered were evaluated toward various areas of the basket. The shifting patterns that produce the maximum reactivity for the 26-assembly and 24-assembly basket configurations are shown in Figure 6.4.4-1 and Figure 6.4.4-2, respectively. Each fuel assembly, tube and disk opening are shifted in the direction indicated by the arrow.

Most Reactive Connecticut Yankee Fuel Configuration Summary

Combining the worst-case fabrication tolerances with the worst-case radial shifting pattern produces the most reactivity configuration for the 26-assembly and 24-assembly baskets. The k_{eff} values calculated for these configurations are 0.9219 for the 26-assembly basket and 0.9313 for the 24-assembly basket (See Tables 6.4.4-12 and 6.4.4-5, respectively). It should be noted that these combinations of fabrication tolerances and mechanical perturbations are highly unrealistic, and unlikely to occur during actual operations. However, by combining conservative assumptions, the most reactive credible configuration is bounded.

6.4.4.2.3 Connecticut Yankee Fuel Assemblies with Missing Fuel Rods

Some of the Connecticut Yankee fuel assemblies are missing individual fuel rods. The exact number and location of these missing rods differ from assembly to assembly. To determine a bounding reactivity for these assemblies, an analysis varying the number of rods removed from the 15 x 15 array is performed. This analysis uses the most reactive 24-assembly basket configuration loaded with Zircaloy clad fuel assemblies enriched to 4.61 wt % ^{235}U . The maximum reactivity resulting from loading assemblies with missing rods in this configuration bounds the maximum reactivity resulting from loading assemblies with the same missing rod geometry into other less reactive configurations such as the 26-assembly basket with 3.93 wt % ^{235}U . Tolerances on basket components, as well as the most reactive radial shifting pattern for the 24-assembly basket configuration, are considered. For each case, all 24 assemblies have the same number and location of missing fuel rods. The results presented in Table 6.4.4-5 show that 24 rods missing from the array represents the bounding number of missing fuel rods for the Connecticut Yankee fuel assemblies. Figure 6.4.4-3 shows the locations of the 24 fuel rods that were removed. In effect, this study has optimized the H/U ratio within all 24 undermoderated fuel assemblies and has resulted in a substantial increase in system reactivity. Hence, loading of assemblies with missing rods will be restricted to the four corner fuel tube locations (openings 1, 4, 23 and 26 as seen in Figure 6.3-3). This configuration results in a system reactivity (k_{eff}) of 0.9327 and corresponds to a change in reactivity, $\Delta k_{\text{eff}} = 0.0014$, which is within the statistics of the code. Therefore, loading of assemblies with missing fuel rods into the four corner fuel tube locations does not significantly affect the overall reactivity of the CY-MPC system.

6.4.4.2.4 Connecticut Yankee Non-Fuel Hardware and Partial Height Flooding

The inventory of the Connecticut Yankee spent fuel pool contains both Reactor Control Cluster Assemblies and Flow Mixers. Either of these components may be placed within a fuel assembly in the CY-MPC. Adding a Reactor Control Cluster Assembly into a fuel assembly displaces moderator in the guide tubes and slightly modifies the reflection material outside the top of the fuel region. As shown in Section 6.4.4.3, the Connecticut Yankee fuel assemblies are undermoderated. Therefore, removing moderator from the lattice decreases reactivity.

When the canister is flooded, the water level inside the canister varies as a function of time. It is not instantaneously filled with water. An additional analysis is performed to evaluate the reactivity effect on the system for the canister cavity flooded only to the top of the active fuel region. This scenario is analyzed to consider the potential increased neutron scattering in the top

end of the canister that may result when water is not present in this region. This scenario is evaluated with the 26-assembly basket configuration, containing the 3.93 wt % ^{235}U , Zircaloy-clad fuel assemblies. A graphical illustration of the partially flooded basket is shown in Figure 6.4.4-9. This case produces a k_{eff} value of 0.9210, and the change in reactivity from the base case ($\Delta k_{\text{eff}} = 0.0009$) is statistically insignificant. Similar results would be expected from the 24-assembly basket configuration.

The partial flooding evaluation demonstrates that removing the moderator above the active fuel region has no significant impact on cask reactivity. Therefore, replacing the water in the guide tubes above the active fuel region and a small portion of water above the upper end fitting with stainless steel from the Reactor Control Cluster Assembly or Flow Mixer would also not have an significant effect on the reactivity of the system.

6.4.4.2.5 Damaged Fuel Assembly Can Evaluation

A number of the fuel assemblies in the Connecticut Yankee spent fuel pool inventory may be classified as damaged, i.e., cladding damage greater than pinhole leaks or hairline cracks. These damaged fuel assemblies must be placed in a damaged fuel can prior to being loaded in the canister. This can has a screened opening in the base plate and the top plate to permit drainage, vacuum drying, and inerting of the can without releasing fissile material from the can. Prior to loading the assemblies with damaged fuel rods into the damaged fuel assembly can, all fuel must be within the confines of the fuel rod cladding. Damaged fuel assembly cans are physically restricted to the 4 oversize fuel tubes in the basket. Assemblies with damaged rods may be placed into a damaged fuel assembly can prior to being loaded into a basket. However, if the damaged fuel assembly cans are first installed in the basket, the damaged fuel should be loaded before loading the remaining (intact) fuel assemblies. This ensures that no fuel material is accidentally dispersed into other regions of the basket that already hold fuel assemblies. Loading the remaining fuel tubes is allowed to commence as long as no fuel material has been dispersed into the canister cavity.

This evaluation considers the CY-MPC 24-assembly basket loaded with 20 of the most reactive fuel assemblies and with 4 damaged fuel assembly cans. Connecticut Yankee 15 x 15 assemblies that have up to 204 damaged fuel rods are analyzed within the damaged fuel assembly can, considering 100% dispersal of the fuel from the rods. All loose fuel in each analysis is modeled as a homogeneous mixture of fuel and water in which the volume fractions of the fuel versus the water are varied from 0-1.0 and vice versa. The clad material of the damaged rods is

conservatively ignored. The bottom of the can is assumed to be in contact with the canister floor. The top of the can lid assembly support ring is assumed to be in contact with the canister lid. Thus, the can cavity height is conservatively extended. Figures 6.4.4-4 through 6.4.4-7 show the model used for the damaged fuel can evaluation. The base case used for reactivity comparisons of these analyses is that of loading the can with an assembly in the most reactive missing rod array containing no damaged or loose fuel. This base case configuration results in a system reactivity, k_{eff} , of 0.9330 with a standard deviation of 0.0008.

First, loose fuel from the damaged fuel rods is distributed between the remaining rods of the assembly in each can. These remaining rods are conservatively assumed to be in the most reactive missing rod array. The results of this analysis, provided in Table 6.4.4.6, show that this scenario slightly decreases the reactivity of the system compared to just placing the most reactive missing rod array within each can. This is due to the fact that adding fuel to the already optimized H/U ratio of the bounding missing rod array reduces the reactivity of the system as this effectively returns the system to an undermoderated state. Second, loose fuel is considered above and below the active fuel region of the most reactive missing rod array within each can. The results of this study, provided in Table 6.4.4.7, show that any possible mixture combination of fuel and water above and below the active fuel region will not significantly increase the reactivity of the system beyond that of the most reactive missing rod array within the can. Loose fuel is also considered to replace all contents of the Connecticut Yankee damaged fuel assembly can in each oversized fuel tube location. The results of this study, provided in Table 6.4.4.8, show that any mixture of fuel and water within the damaged fuel assembly can cavity will not significantly increase the reactivity of the system beyond that of placing an assembly having the most reactive missing rod array within each can.

The most reactive damaged fuel configuration occurs when the fully flooded 24-assembly basket configuration is loaded with 4 damaged fuel assembly cans, with each fully flooded can containing an assembly having the most reactive missing rod array. This configuration results in a system reactivity of 0.9330. This corresponds to a change in reactivity, $\Delta k_{\text{eff}} = 0.0017$, which is considered to be within the statistics of the code when compared to loading the 24-assembly basket configuration with 24 Westinghouse Vantage 5H assemblies. Loading the 26-assembly basket with 4 damaged fuel cans containing Zircaloy-clad fuel assemblies enriched up to 3.93 wt % ^{235}U or stainless steel clad fuel assemblies, either type with up to 204 damaged fuel rods, would behave similarly. Therefore, loading the basket with damaged fuel assembly cans containing assemblies with up to 204 damaged rods does not affect the overall reactivity of the CY-MPC

system as long as the enrichment of the fuel within each can is limited to that of the basket configuration in which it is loaded.

6.4.4.2.6 CY Reconfigured Fuel Assembly Evaluation

The Connecticut Yankee spent fuel pool inventory includes a number of individual fuel rods that have been removed from their host fuel assemblies and may be intact or damaged. The CY reconfigured fuel assembly is designed to accommodate these individual fuel rods, or other fuel debris. The key physical dimensions and limits on the fuel material to be stored are listed in Table 6.2-5. The CY reconfigured fuel assembly restricts the fuel rods to a 10 x 10 array of tubes that are captured between top and bottom tie-plates. The tie-plates have a smaller drain line at each tube. There is a screened drain reservoir below the bottom tie-plate. The analysis of the CY reconfigured fuel assembly assumes that the fuel rods contain fuel with an initial enrichment of 4.61 wt % ^{235}U . For conservatism, the fuel rod cladding is not modeled within each individual tube. The CY reconfigured fuel assembly is designed to be physically restricted to the four corner fuel tube locations. However, the analysis conservatively evaluates the assemblies in all 26 basket locations to demonstrate that the CY reconfigured fuel assembly design has a lower reactivity than either of the design basis intact fuel assembly configurations.

Four individual studies are performed to demonstrate that the CY reconfigured fuel assembly is bounded by the design basis intact fuel evaluation. The first study evaluates the reactivity of the CY reconfigured fuel assembly as a function of the fuel pellet diameter within the tubes as shown in Table 6.4.4-9. The fuel pellet diameter inside each tube is decreased from the 1.2814 cm maximum. The reported variation in reactivity between diameters of 1.2814 cm and 1.27 cm is statistically insignificant. Therefore, the largest diameter is assumed to be the most reactive, as this maximizes the amount of fissile material in the system. The second study considers variations in water moderator density within the flooded canister, including within the CY reconfigured fuel assembly. Results for this moderator density study are shown in Table 6.4.4-10 and demonstrate that the effective full density moderator condition is the most reactive, and that the reactivity of the CY reconfigured fuel assembly is well below that of the design basis intact fuel assembly. The third study considers a homogeneous volume fraction study of a fuel-water mixture within each reconfigured fuel assembly tube. Results for this evaluation are shown in Table 6.4.4-11. Excluding the statistically insignificant variations in reactivity that occur for water volume fractions below 10%, the most reactive configuration is for the individual tubes completely filled with fuel material. The fourth study considers the reactivity impact of fuel debris in the drain lines and drain reservoir in the bottom end fitting, as depicted

in Figure 6.4.4-8. In this figure, the drain lines in the bottom tie-plate are conservatively considered to be the same size as the tube. The results of this case, $k_{\text{eff}} = 0.9079$, show that modeling the drain tubes beneath each of the tubes in the 10 x 10 array and the drain reservoir with solid fuel material surrounded by water, is the most reactive CY reconfigured fuel assembly configuration.

Comparison of the most reactive CY reconfigured fuel assembly case to either of the design basis intact fuel assembly cases demonstrates that the reactivity of the CY reconfigured fuel assembly is bounded by that of the design basis fuel assemblies. As an additional conservative margin, CY reconfigured fuel assemblies can be loaded only in the 4 corner locations of the fuel basket.

6.4.4.3 Normal Conditions – Connecticut Yankee Fuel

Connecticut Yankee fuel assemblies will be sealed inside a canister that is welded shut. Consequently, the canistered fuel is dry under normal conditions of loading and transport. Evaluation of normal conditions, therefore, requires the moderator density to be ~ 0.0 g/cc (dry). The resulting k_{eff} for normal conditions of transport are 0.3555 ± 0.0008 for the 26-assembly basket configuration and 0.3715 ± 0.0008 for the 24-assembly basket configuration as shown in the last row of Table 6.4.4-12. For conservatism, the transport cask criticality analysis also considers a flooded normal condition. The maximum reactivity for this scenario results in a k_{eff} of 0.9313 ± 0.0008 for the 24-assembly basket configuration and a k_{eff} of 0.9219 ± 0.0008 for the 26-assembly basket configuration.

Including statistical uncertainties, all results for the normal condition are below the 0.9425 upper subcritical limit defined in Section 6.5.2. Thus, compliance with 10 CFR 71.55 (b) and (d) as well as 10 CFR 71.75 (a) is demonstrated.

6.4.4.4 Hypothetical Accident Conditions – Connecticut Yankee Fuel

Criticality analysis of the hypothetical accident conditions includes variations in moderator density from 1.0 g/cc to ~ 0.0 g/cc (dry). The cask, canister and fuel are considered to be fully moderated as described in Section 6.1.2. The moderator density studies, provided in Tables 6.4.4-12 and 6.4.4-13, show that optimum moderation occurs at full density and results in a system reactivity, $k_{\text{eff}} \pm \sigma$, of 0.9313 ± 0.0008 for the 24-assembly basket and a k_{eff} of 0.9219 ± 0.0008 for the 26-assembly basket.

A case discretely modeling a partially flooded canister loaded with four fully flooded damaged fuel cans was also evaluated. The geometry of this case is identical to the most reactive damaged fuel can configuration documented in Section 6.4.4.2.5 with the following exception. The canister cavity is partially flooded with full density moderator to a height approximately equal to the middle of the active fuel. The four damaged fuel cans are hypothetically assumed to remain fully flooded along the entire axial extent of each can. This geometry bounds a postulated uneven drain down condition for a canister containing damaged fuel cans. Comparison of the calculated reactivity of this configuration, $k_{\text{eff}} \pm 2\sigma = 0.9259 \pm 0.0008$, to the calculated reactivity of the same basket loading with full density moderator completely filling the canister cavity, $k_{\text{eff}} \pm 2\sigma = 0.9330 \pm 0.0008$, demonstrates that the full canister flooding scenario is bounding. Similar results would be expected from the 26-assembly basket configuration.

Axial shifting of the contents of the CY-MPC system is considered as a result of the top end impact accident condition. A bounding hypothetical fuel-shifting scenario is considered. This scenario conservatively shifts all fuel rods to the top of each assembly. The fuel within these rods is assumed to shift into half the height of the plenum, and each assembly is shifted up until it is in contact with the lid. The conservatively toleranced basket is assumed to remain in contact with the canister floor. The MONK8a criticality analysis sequence of the ANSWERS software code evaluated the change in reactivity of the system as a result of this scenario to a statistical uncertainty of 0.0004.

As a result of this shifting scenario, some of the active fuel protrudes beyond the top of the BORAL. The Connecticut Yankee fuel assembly dimensions that are used to determine this height are presented in Table 6.2-4. The height of active fuel that protrudes beyond the top of the BORAL sheets is calculated by taking the maximum possible height from the canister lid to the top of the BORAL and subtracting the minimum distance from the top of the assembly to the top of the active fuel.

Top End Impact Evaluation

Analysis of the hypothetical top end impact accident condition is performed using the most reactive 24-assembly basket configuration. Structural evaluations show that in the transport cask hypothetical top end impact event, PWR top nozzles do not deform. Therefore, the hypothetical shifting condition of a CY assembly with the axial dimensions presented in Table 6.2-4 results in 1.905 inches of active fuel protruding beyond the top of the BORAL. To provide extra margin, this analysis conservatively considers 4 inches active fuel protruding beyond the top of the BORAL by reducing the BORAL sheet length by 2.095 inches. This condition establishes a

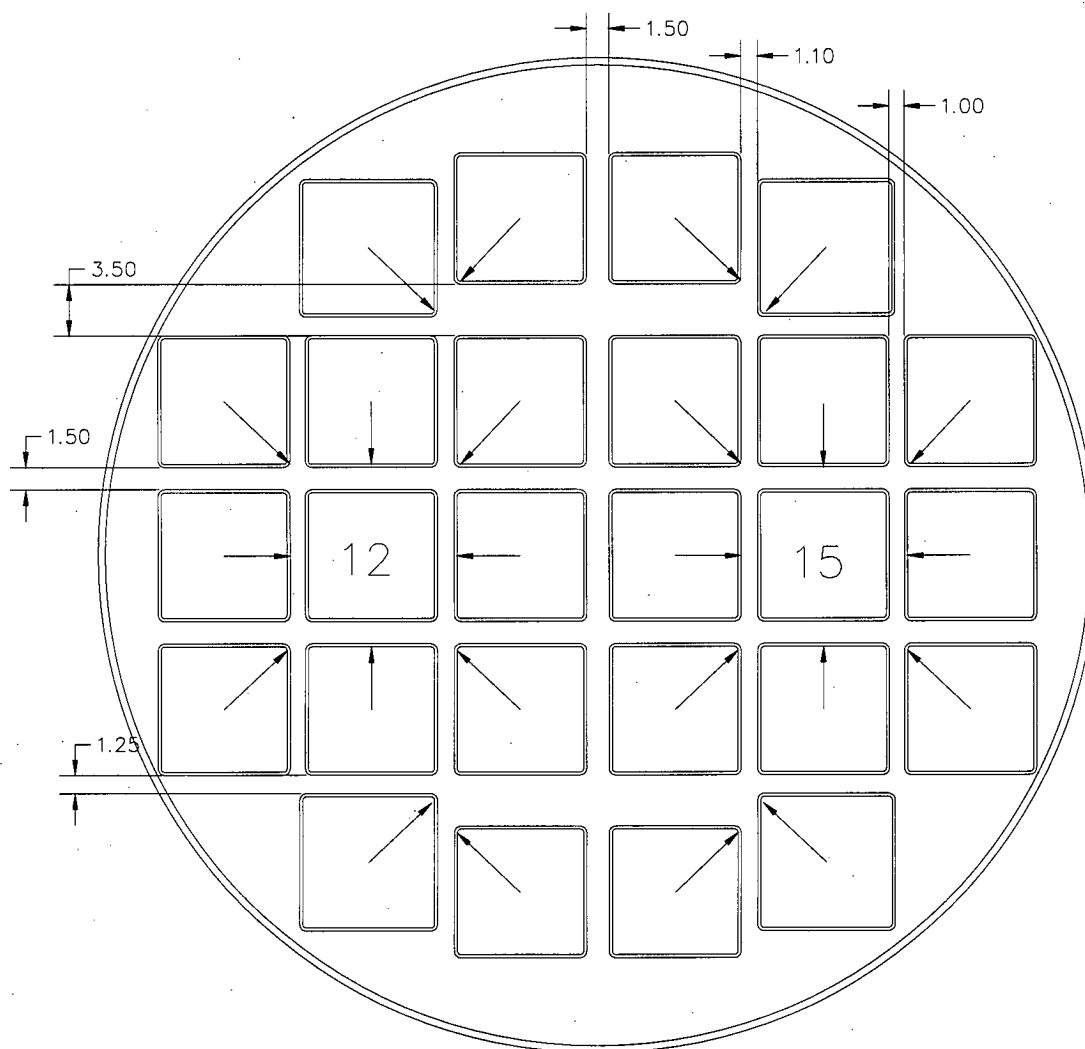
transport cask top end impact system reactivity (k_{eff}) of 0.9327 upon exposing 4 inches of active fuel above the BORAL, as shown in Figure 6.4.4-10.

Bottom End Impact Evaluation

The maximum distance from the canister floor to the bottom of the BORAL occurs when the conservatively toleranced basket components are shifted up towards the canister lid. For the CY-MPC, this distance is limited to 5.41 inches. All Connecticut Yankee fuel types to be loaded have rod end caps, tie plates and/or bottom nozzle components that do not deform to a total height of less than 1.41 inches. Therefore, the top end impact event, which exposes 4 inches of PWR fuel, bounds the bottom end impact condition.

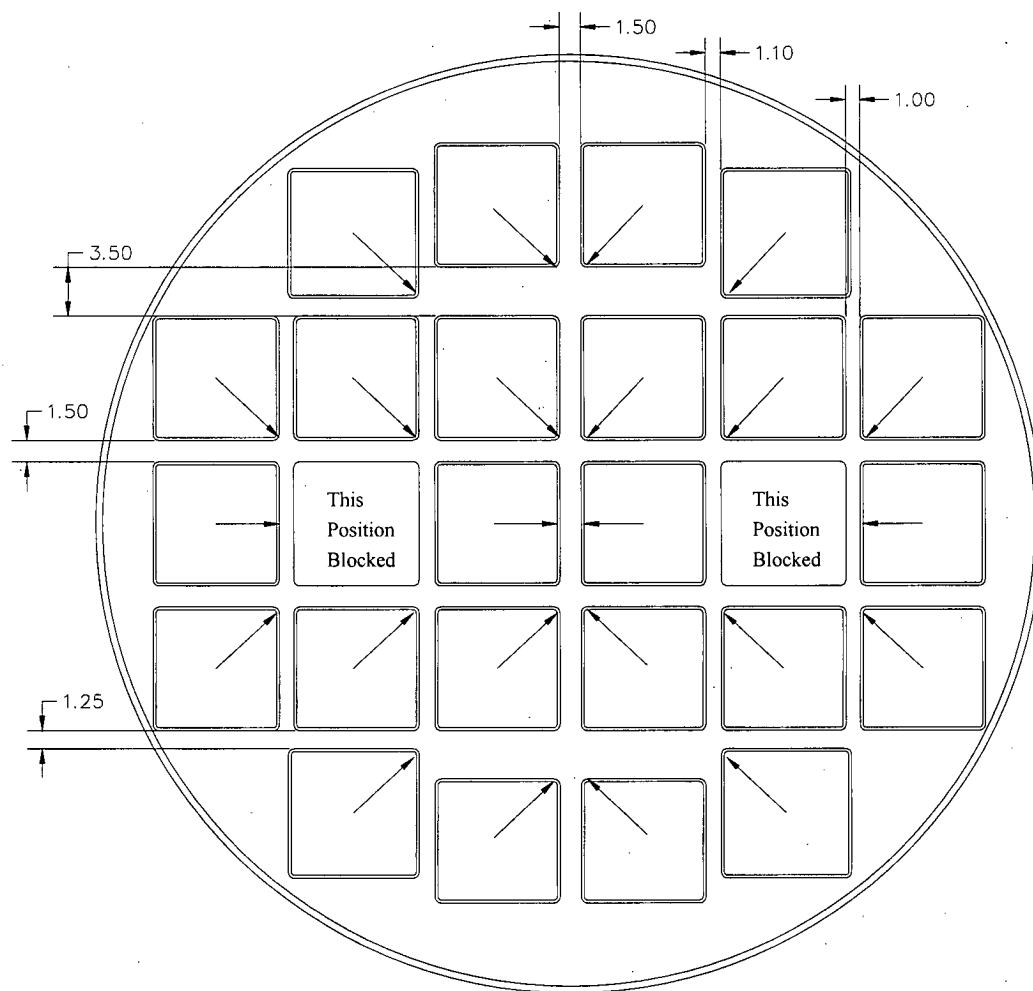
Given that the bounding end impact event does not significantly affect the reactivity of the system, poison sheet coverage is adequate for all allowed contents of the CY-MPC system. The maximum system reactivity, including statistical uncertainties, optimum moderation, mechanical perturbations, tolerances, shifting, and the hypothetical accident conditions is 0.9329. The MONK8a output for this case is shown in Figure 6.7-9. Including statistical uncertainties, all results for the hypothetical accident conditions of transport are well below the 0.9425 upper subcritical limit. Thus, compliance with 10 CFR 71.55 (e) is demonstrated.

Figure 6.4.4-1 CY-MPC Maximum Reactivity Shifting Pattern – 26-Assembly Basket



(Units = inches, Not drawn to scale)

Figure 6.4.4-2 CY-MPC Maximum Reactivity Shifting Pattern – 24-Assembly Basket



(Units = inches, Not drawn to scale)

Figure 6.4.4-3 CY-MPC Most Reactive Missing Fuel Rod Geometry

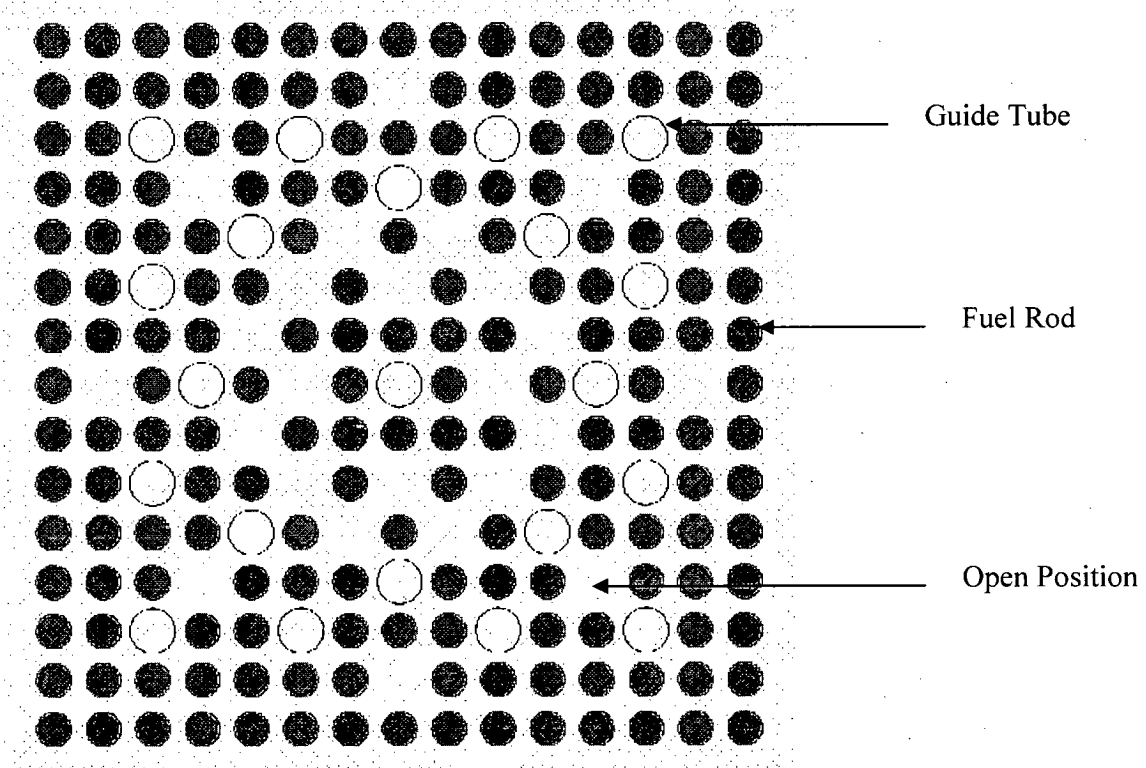


Figure 6.4.4-4 Shifted CY Damaged Fuel Can and Most Reactive Missing Rod Array

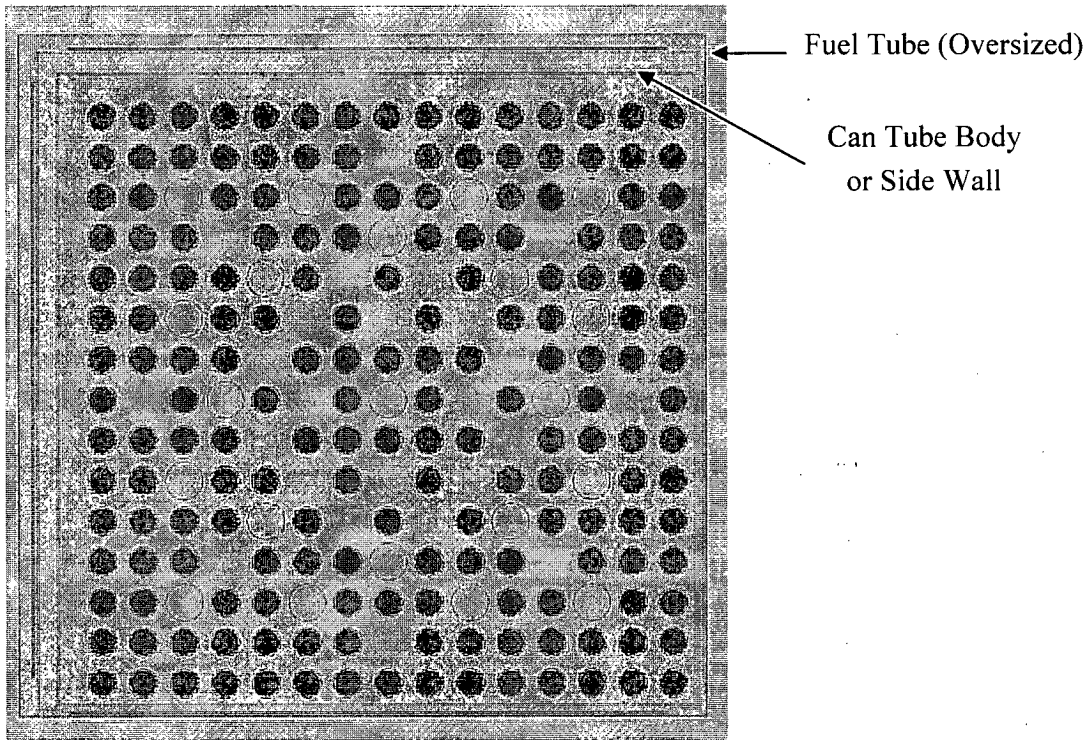


Figure 6.4.4-5 CY-MPC Model Geometry Below Active Fuel

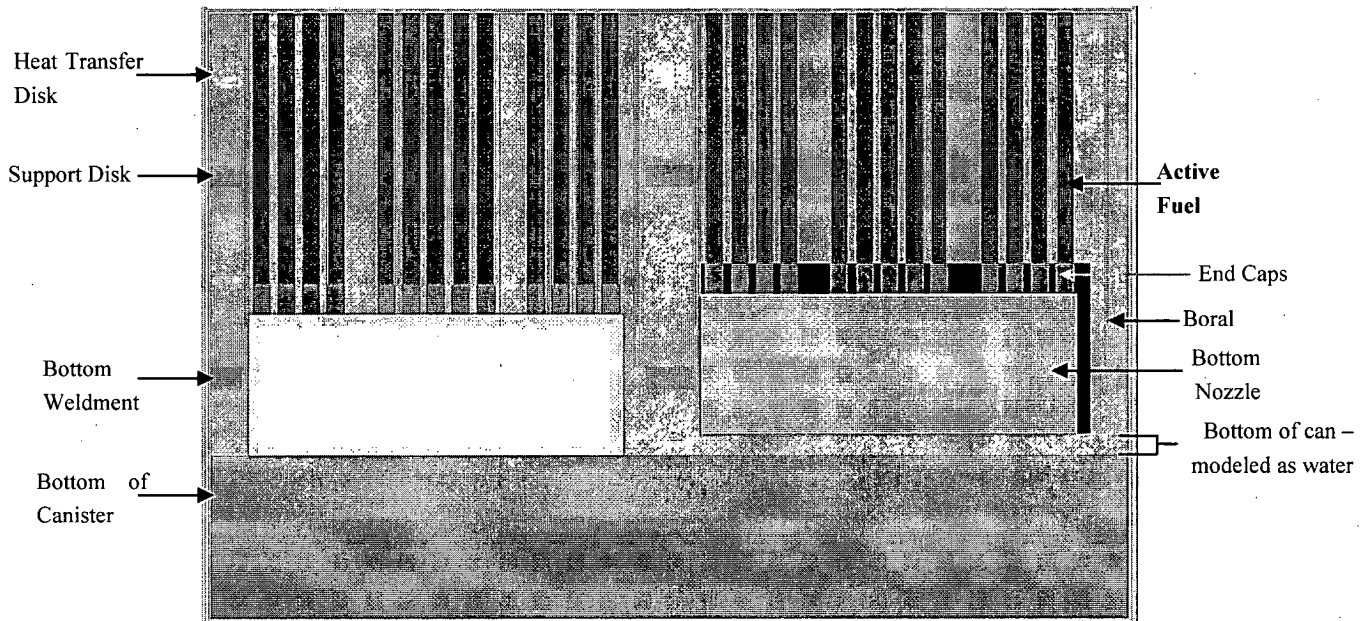


Figure 6.4.4-6 CY-MPC Model Geometry Above Active Fuel

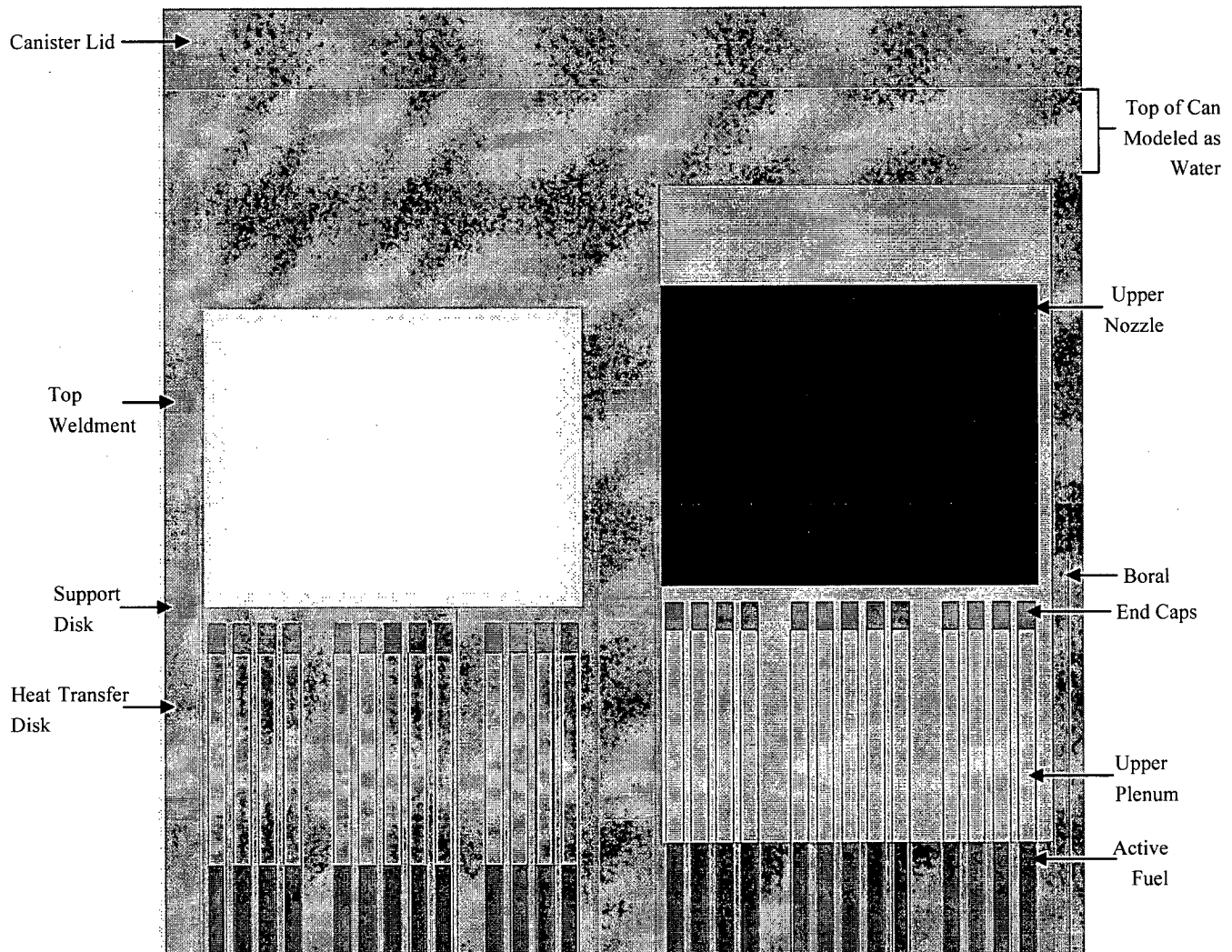


Figure 6.4.4-7 CY-MPC Model Slice of Two Corner Tubes with Damaged Fuel Cans and Two Middle Tubes

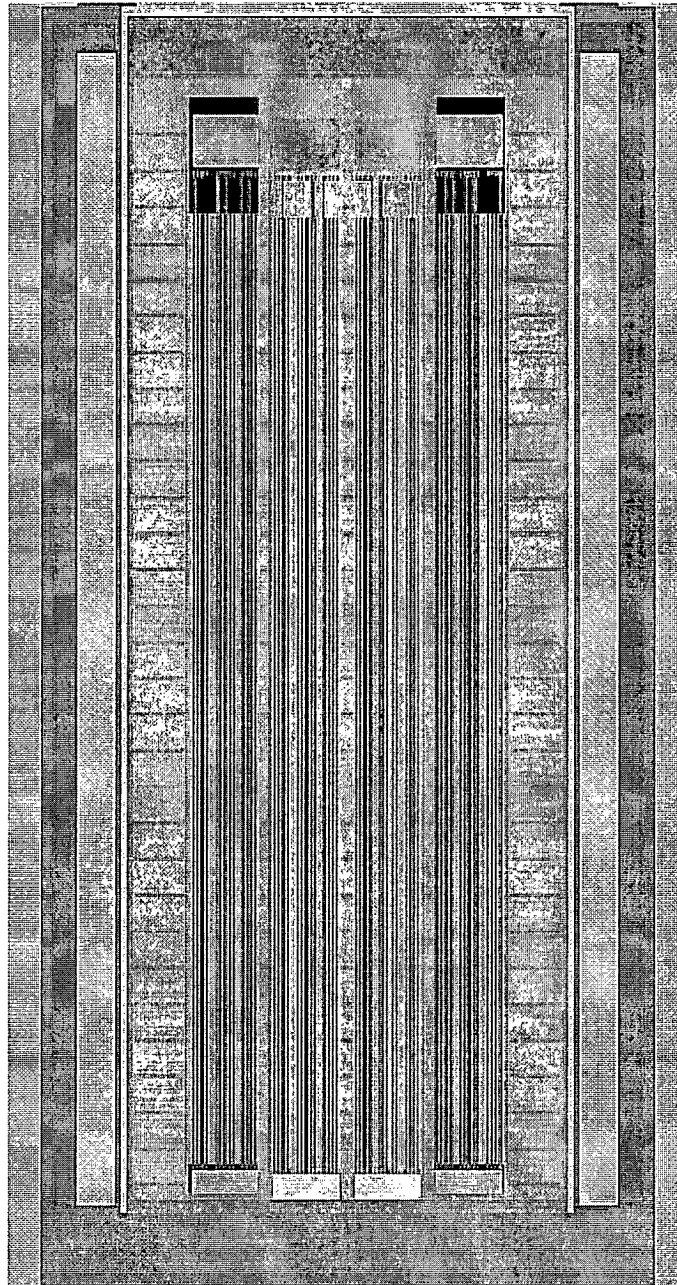


Figure 6.4.4-8 CY Reconfigured Fuel Assembly - Axial Model Configuration with Fuel in Drain Reservoir

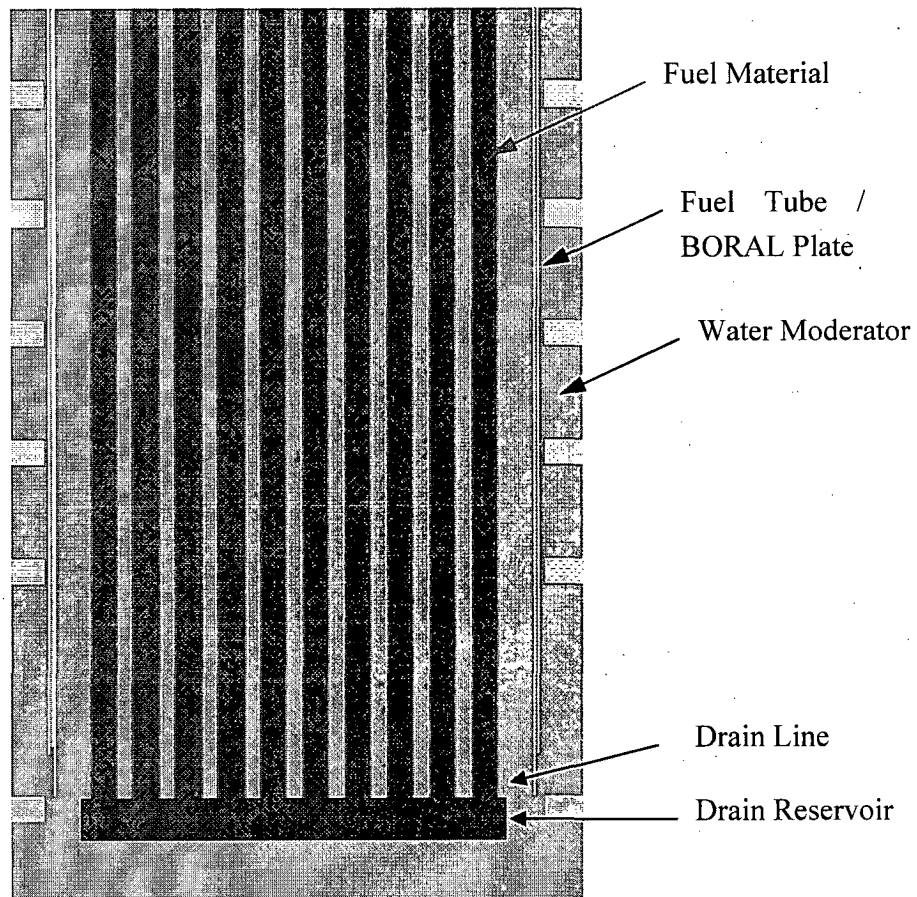


Figure 6.4.4-9 CY-MPC Partial Flooding Model – Top of Active Fuel Region

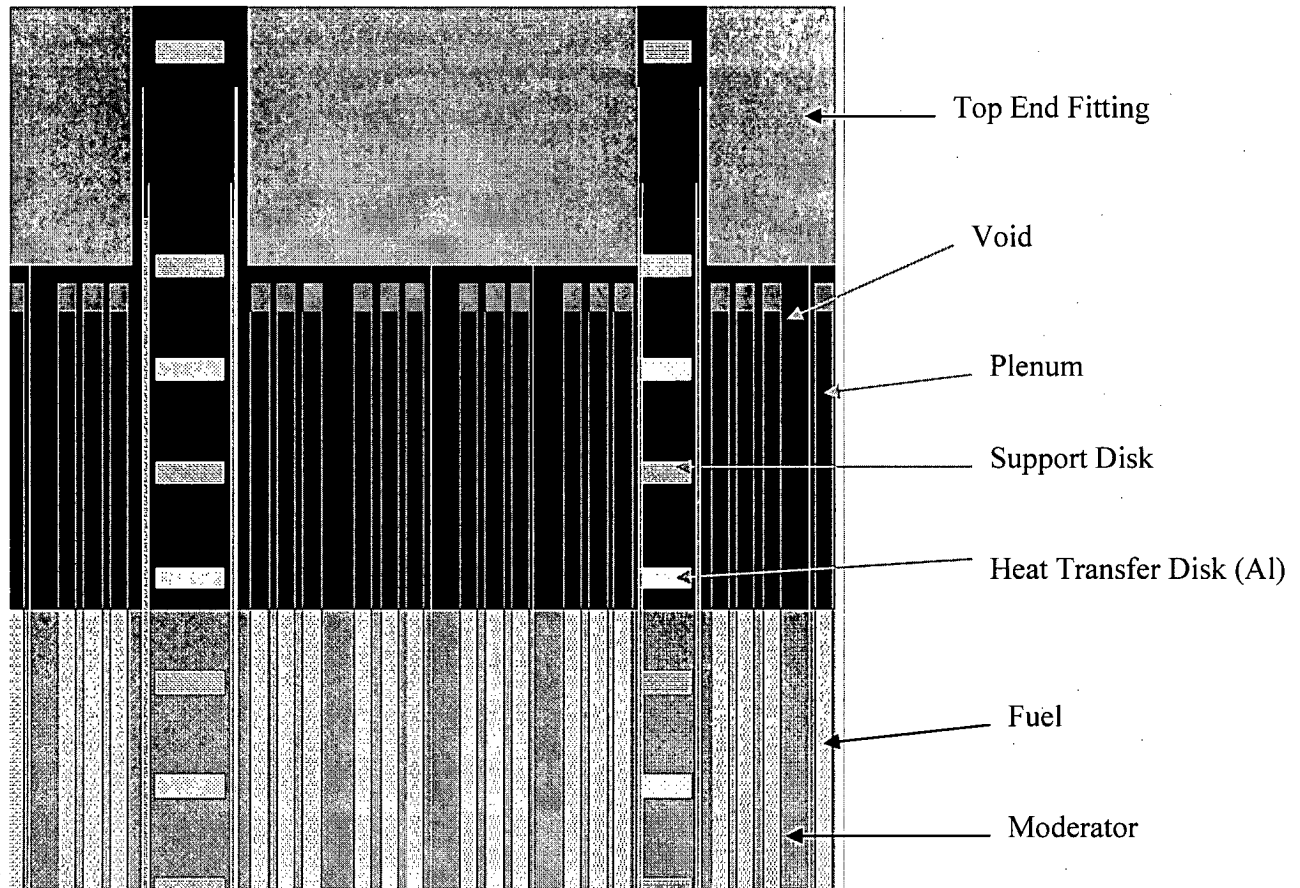


Figure 6.4.4-10 CY-MPC Top Impact Model – Top of the Shifted Active Fuel Region

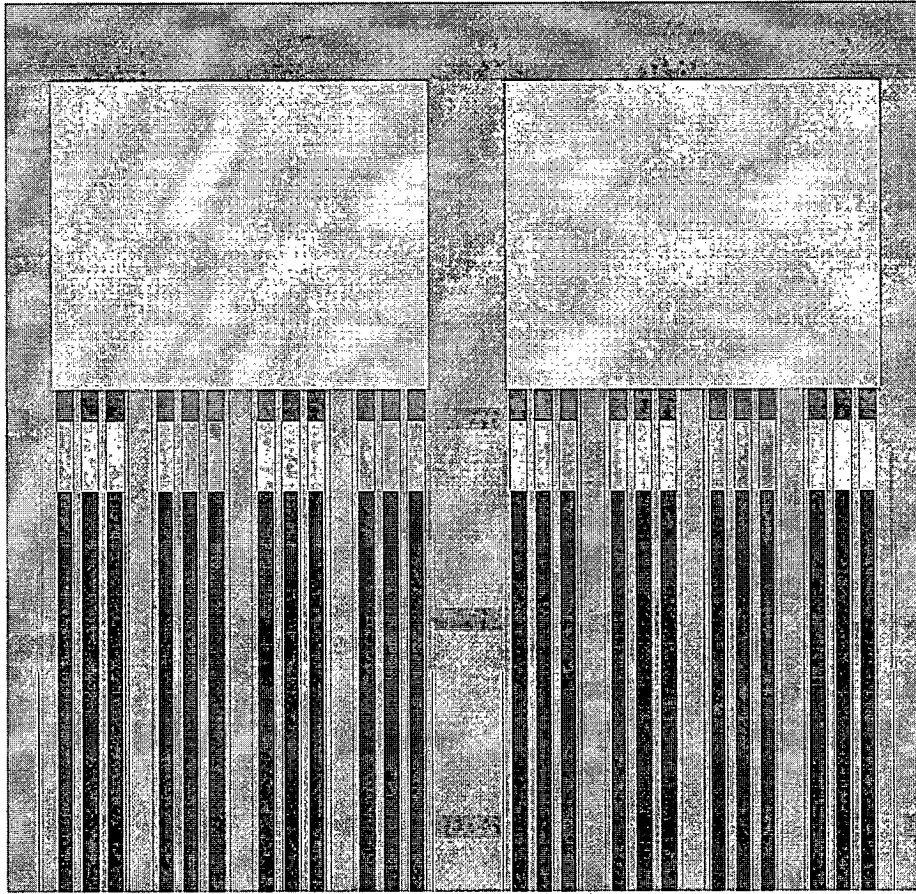


Table 6.4.4-1 Connecticut Yankee Most Reactive Fuel Assembly Evaluation

	Enrichment (wt % ²³⁵ U)	k _{eff} - Flooded Gap		
		Min Pellet OD	Mid Pellet OD	Max Pellet OD
Stainless Steel	4.03	0.8898	0.8903	0.8904
Zircaloy	3.42	0.8840	0.8834	0.8824
Zircaloy	3.93	0.9073	0.9077	0.9064
Zircaloy	4.61	0.9454	0.9445	0.9460

Table 6.4.4-2 CY-MPC Fabrication Tolerance Evaluation

Model	k _{eff}	Δ k _{eff}
Base Case	0.9460	n/a
BORAL sheet minus tolerances	0.9479	0.0019
Disk opening center minus tolerance	0.9456	-0.0004
Disk opening center plus tolerance	0.9454	-0.0006
Disk opening minus tolerance	0.9457	-0.0003
Disk opening plus tolerance	0.9434	-0.0026
Tube opening minus tolerance	0.9413	-0.0047
Tube opening plus tolerance	0.9500	0.0040
Combined BORAL sheet and tube opening tolerances	0.9494	0.0034

Note: The Base Case is 26 Zircaloy clad fuel assemblies with 4.61 wt % ²³⁵U, maximum pellet diameter, centered in the tube openings.

Table 6.4.4-3 CY-MPC Mechanical Perturbation Evaluation – 26-Assembly Basket

Configuration	k_{eff}	Δk_{eff}
Base Case – No shift	0.9064	NA
Toward Openings 12 and 15	0.9153	0.0089
Toward Openings 13 and 14	0.9136	0.0072
9, 10, 15, 16 towards each other; 11, 12, 17, 18 towards each other; remaining openings towards 13 and 14	0.9129	0.0065
6, 12, 18 towards 11; 9, 15, 21 towards 16; remaining openings towards 13 and 14	0.9120	0.0056
2, 6, 12, 18, 24 towards 11; 3, 9, 15, 21, 25 towards 16; remaining openings towards 13 and 14	0.9119	0.0055
4 peripheral clusters targeting 1.1 inch gaps center row also targeting 1.1 inch gaps	0.9096	0.0032
6 peripheral clusters targeting 1.0 inch gaps; remaining openings (which are adjacent to 1.5 inch gap) towards 13 and 14	0.9089	0.0025
4 peripheral clusters targeting 1.1 inch gaps 3 clusters in center row	0.9081	0.0011

Notes:

1. Fuel position numbers are as shown in Figure 6.3-3.
2. The Base Case is 26 Zircaloy clad fuel assemblies with 3.93 wt % ^{235}U , maximum pellet diameter, maximum disk opening size, with the fuel tubes and disk openings shifted as specified.

Table 6.4.4-4 CY-MPC Mechanical Perturbation Evaluation – 24-Assembly Basket

Configuration	k_{eff}	Δk_{eff}
Base case - No shift	0.9197	NA
Toward Openings 13 and 14	0.9275	0.0078
4 peripheral clusters targeting 1.1 inch gaps; remaining openings towards 13 and 14	0.9261	0.0064
4 peripheral clusters targeting 1.1 inch gaps; remaining openings towards 1.5 inch gap	0.9260	0.0063
4 peripheral clusters targeting 1.0 inch gaps; remaining openings towards 13 and 14	0.9257	0.0060
Openings surrounding 12 or 15 towards 12 or 15, respectively; remaining openings towards 13 and 14	0.9226	0.0029
Openings surrounding 12 or 15 towards 12 or 15, respectively, remaining openings in 4 clusters targeting 1.1 inch gaps	0.9219	0.0022

Note:

1. Fuel position numbers are as shown in Figure 6.3-3.
2. The Base Case is 24 Zircaloy clad fuel assemblies with 4.61 wt % ^{235}U , maximum pellet diameter, maximum disk opening size, with the fuel tubes and disk openings shifted as specified.

Table 6.4.4-5 CY-MPC Missing Fuel Rod Evaluation Results

Number of Missing Rods	k_{eff}	Δk_{eff}
0	0.9313	NA
4	0.9396	0.0083
8	0.9376	0.0063
12	0.9401	0.0088
16	0.9413	0.0100
20	0.9441	0.0128
24	0.9464	0.0151
28	0.9444	0.0131
32	0.9439	0.0126
36	0.9463	0.0150
40	0.9453	0.0140
44	0.9399	0.0086
48	0.9396	0.0083
56	0.9384	0.0071

Note: The 0 missing rods case is 24 Zircaloy clad fuel assemblies with 4.61 wt % ^{235}U , maximum tube opening, minimum BORAL width, maximum disk opening size, with the fuel, tubes and disk openings shifted towards openings 13 and 14.

Table 6.4.4-6 Mixture of Damaged Fuel and Water within the Active Fuel Region of Intact Rods in Each Can

V_f of UO_2	k_{eff}	Δk_{eff}
0.00	0.9330	NA
0.20	0.9300	-0.0030
0.40	0.9276	-0.0054
0.60	0.9257	-0.0073
0.80	0.9284	-0.0046
1.00	0.9263	-0.0067

Table 6.4.4-7 Mixture of Damaged Fuel and Water outside the Active Fuel Region of Intact Rods in Each Can

V_f of UO_2	k_{eff}	Δk_{eff}
0.00	0.9330	NA*
0.20	0.9337	0.0007
0.40	0.9333	0.0003
0.60	0.9320	-0.0010
0.80	0.9310	-0.0020
1.00	0.9328	-0.0002

* Base Case with 0.00 UO_2 in water between rods from Table 6.4.4-6.

Table 6.4.4-8 Mixture of Damaged Fuel and Water Replacing Contents of Each Can

V_f of UO_2	k_{eff}	Δk_{eff}
NA*	0.9330	NA
0.00	0.9258	-0.0072
0.20	0.9336	0.0006
0.40	0.9281	-0.0049
0.60	0.9288	-0.0042
0.80	0.9268	-0.0062
1.00	0.9266	-0.0064

* Base case with 0.00 UO_2 in water between rods from Table 6.4.4-6.

Table 6.4.4-9 CY Reconfigured Fuel Assembly – Reactivity as a Function of Fuel Pellet Diameter

Fuel Diameter (cm)	k_{eff}
1.2814	0.9057
1.270	0.9059
1.250	0.9045
1.245	0.9049
1.240	0.9019
1.235	0.9023
1.230	0.9036
1.225	0.9009
1.200	0.8994
1.100	0.8812
1.000	0.8569

Table 6.4.4-10 CY Reconfigured Fuel Assembly – Reactivity as a Function of Moderator Density Variation

Moderator Density (g/cc)	k _{eff}
0.9982	0.9057
0.975	0.8990
0.95	0.8949
0.90	0.8800
0.85	0.8699
0.80	0.8587
0.60	0.8022
0.40	0.7245
0.20	0.5979
0.000001 (dry cask)	0.3756

Table 6.4.4-11 CY Reconfigured Fuel Assembly – Reactivity as a Function of Variations in Fuel/Water Homogeneous Mixture Volume Fraction

Water Volume Fraction	k_{eff}
0.00	0.9057
0.05	0.9032
0.06	0.9047
0.07	0.9059
0.08	0.9020
0.09	0.9020
0.10	0.9032
0.15	0.8986
0.20	0.8962
0.25	0.8931
0.30	0.8873
0.35	0.8831
0.40	0.8695
0.45	0.8618
0.50	0.8466
0.55	0.8305
0.60	0.8119
0.65	0.7846
0.70	0.7513
0.75	0.7095

Table 6.4.4-12 CY-MPC Transfer Cask Analyses – Reactivity as a Function of Moderator Density Variations

Water Density	26-Assembly Basket Configuration	24-Assembly Basket Configuration
(g/cc)	k_{eff}	k_{eff}
0.9982	0.9219	0.9313
0.95	0.9067	0.9173
0.90	0.8930	0.9006
0.85	0.8769	0.8818
0.80	0.8595	0.8622
0.60	0.7818	0.7767
0.40	0.6832	0.6729
0.20	0.5520	0.5500
0.000001 ~ 0.0	0.3555	0.3715

Table 6.4.4-13 CY Damaged Fuel Can Analyses – Reactivity as a Function of Moderator Density Variation

Moderator Density		k_{eff}	Δk_{eff}
Canister	Can		
1.00	1.00	0.9313	NA
1.00	0.00	0.9253	-0.0060
1.00	0.20	0.9263	-0.0050
1.00	0.40	0.9271	-0.0042
1.00	0.60	0.9264	-0.0049
1.00	0.80	0.9282	-0.0031
0.00	1.00	0.9149	-0.0164
0.20	1.00	0.8744	-0.0569
0.40	1.00	0.8692	-0.0621
0.60	1.00	0.8765	-0.0548
0.80	1.00	0.8892	-0.0421

6.5 Critical Benchmark Experiments

This section provides the validation of the computer codes used in the criticality evaluation of the NAC-STC directly loaded and the Yankee-MPC and CY-MPC canistered fuel systems. This validation is required by the criticality safety standard ANSI/ANS-8.1 [10].

Section 6.5.1 describes the methodology, computer program and cross section libraries used, the experimental data, the areas of applicability and the bias and margins of safety applied in the SCALE 4.3 analysis package used in the evaluation of the NAC-STC directly loaded and Yankee-MPC canistered fuel configurations. Section 6.5.2 describes the MONK8a code applied in the analysis of the CY-MPC canistered fuel configuration.

ANSI/ANS-8.17 prescribes the criteria to establish subcritical safety margins. This criteria is:

$$k_s \leq k_c - \Delta k_s - \Delta k_c - \Delta k_m \quad (1)$$

where,

k_s = the calculated allowable maximum multiplication factor, k_{eff} , of the system being evaluated for all normal or credible abnormal conditions or events.

k_c = the mean k_{eff} that results from the calculation of the benchmark criticality experiments using a particular calculational method. If the calculated k_{eff} for the criticality experiments exhibit a trend with a parameter, then k_c shall be determined by extrapolation on the basis of a best fit to the calculated values. The criticality experiments used as benchmarks in computing k_c should have physical compositions, configurations, and nuclear characteristics (including reflectors) similar to those of the system being evaluated.

Δk_s = an allowance for

- (a) statistical or convergence uncertainties, or both, in the computation of k_s ,
- (b) material and fabrication tolerances, and
- (c) geometric or material representations used in the computational method.

Δk_c = a margin for uncertainty in k_c which includes allowance for

- (a) uncertainties in the critical experiments,
- (b) statistical or convergence uncertainties, or both, in the computation of k_c ,
- (c) uncertainties due to extrapolation of k_c outside the range of experimental data, and
- (d) uncertainties due to limitations in the geometrical or material representations used in the computational method.

Δk_m = an arbitrary margin to ensure the subcriticality of k_s

The various uncertainties are combined statistically if they are independent. Correlated uncertainties are combined additively. The above equation (1) can be rewritten as:

$$k_s \leq 1 - \Delta k_m - \Delta k_s - (1 - k_c) - \Delta k_c \quad (2)$$

Noting that the NRC requires a 5% subcriticality margin ($\Delta k_m = 0.05$) and the definition of the bias ($\beta = 1 - k_c$), this equation can then be written as:

$$k_s \leq 0.95 - \Delta k_s - \beta - \Delta \beta \quad (3)$$

where $\Delta \beta = \Delta k_c$. Thus, k_s (the maximum allowable value for k_{eff}) must be below 0.95 minus the bias, uncertainties in the bias and uncertainties in the system being analyzed (i.e., Monte Carlo, mechanical and modeling). This is an upper safety limit criterion often used in the DOE criticality safety community.

Alternatively, this equation can be rewritten applying the bias and uncertainties to the k_{eff} of the system being analyzed as:

$$k_s \equiv k_{eff} + \Delta k_s + \beta + \Delta \beta \leq 0.95 \quad (4)$$

In equation 4, k_{eff} replaces k_s , and k_s has been redefined as the effective multiplication factor of the system being analyzed, including the method bias and all uncertainties. This is a maximum calculated k_{eff} criteria often used in LWR spent fuel storage and transport analyses.

β and $\Delta \beta$ are evaluated in Section 6.5.1 for the NAC-STC directly loaded and Yankee-MPC canistered fuel configurations and in Section 6.5.2 for the CY-MPC fuel configuration.

6.5.1 Benchmark Experiments and Applicability for CSAS25

The criticality safety method applied to the NAC-STC directly loaded and Yankee-MPC canistered fuel configurations is CSAS25 embedded in SCALE Version 4.3 for the PC. CSAS25 includes the SCALE Material Information Processor, BONAMI, NITAWL-II, and KENO-Va. The Material Information Processor generates number densities for standard compositions, prepares geometry data for resonance self-shielding, and creates data input files for the cross section processing codes. The BONAMI and NITAWL-II codes are used to prepare a resonance-corrected cross section library in AMPX working format. The KENO-Va code calculates the model k_{eff} using Monte Carlo techniques. The 27-group ENDF/B-IV neutron cross section library is used in this validation.

6.5.1.1 Description of Experiments

Sixty-three critical experiments were selected: nine Babcox and Wilcox 2.46 wt % ^{235}U fuel storage [11]; ten Pacific Northwest Laboratory 4.31 wt % ^{235}U lattice [12]; twenty-one PNL 2.35 and 4.31 wt % ^{235}U with metal reflectors [13, 14]; twelve PNL flux trap [10, 15]; and, eleven Valduc Critical Mass Laboratory 4.74 wt % ^{235}U , some involving moderator density variations [16, 17]. These experiments span a range of fuel enrichments, fuel rod pitches, neutron absorber sheet characteristics, shielding materials and geometries that are typical of LWR fuel in a cask.

6.5.1.2 Applicability of Experiments

All of the experiments chosen in this validation are applicable to either PWR, including Yankee Class fuel, or to BWR fuel. Fuel enrichments have covered a range from 2.35 up to 4.74 wt % ^{235}U typical of the fuels presently used. The experiment fuel rod and pitch characteristics are within the range of standard PWR or BWR fuel rods (i.e., pellet diameters from 0.78 to 1.2 cm, rod diameters from 0.95 to 1.88 cm and pitches from 1.26 to 1.87 cm). This is particularly true of the Valduc Critical Mass Laboratory (PWR rod type) and Babcock and Wilcox experiments (BWR rod type). The H/U volume ratios of the experimental fuel arrays are within the range of PWR fuel assemblies (1.6 to 2.32) and BWR fuel assemblies (1.6 to 1.9).

In the case of the Yankee Class fuel, the majority of the Zircaloy clad fuel has an enrichment below 4.03 wt % ^{235}U , and the stainless steel clad fuel is 4.97 wt % ^{235}U , just outside the experimental range. However, the stainless steel clad fuel is much less reactive than the Zircaloy clad and is not limiting. Also, in the case of the Yankee Class fuel, the pellet diameter varies from 0.747 to 0.789

cm, the rod outside diameter varies from 0.864 to 0.927 cm and the pitch varies from 1.07 to 1.20 cm, and the resultant H/U volume ratio varies from 1.28 to 1.57. These fuel parameters are all slightly outside of the range of experiments, but given the lack of statistically significant trends as demonstrated in Figures 6.5.1-2 through 6.5.1-7, confidence in criticality prediction by extrapolation to the Yankee fuel parameter is still high.

Experiments covered the geometry and neutron absorber sheet arrangements typical of the NAC-STC directly loaded and Yankee-MPC canister basket designs. This included a flux trap gap spacing of 3.81 cm such as in the NAC-STC directly loaded basket, and gap spacing as low as 1.91 cm such as in the canister basket used for the Yankee Class fuel. The ^{10}B neutron absorber loadings are also typical of these basket designs (0.005 to 0.025). The experiments covered the influence of water and metal reflector regions, including steel and lead, which would be present in storage and transport cask shielding.

Confidence in predicting criticality, including bias and uncertainty, has been demonstrated for spent fuel with enrichments up to 4.74 wt % ^{235}U and, based on the lack of a significant trend with enrichment, confidence in extrapolation up to 4.97 wt % ^{235}U is high. Confidence in predicting criticality has been demonstrated for storage and transport arrays using flux trap or single neutron absorber sheet or simple spacing criticality control. Confidence in predicting criticality has been demonstrated for spent fuel storage and transport arrays next to water and metal reflector regions.

6.5.1.3 Results of Benchmark Calculations

The k-effective results for the experiments are shown in Table 6.5.1-1 and a frequency distribution plot is provided in Figure 6.5.1-1. Five sets of cases are presented: Set 1 - B&W, Set 2 - PNL lattice, Set 3 - PNL reflector, Set 4 - PNL flux trap and Set 5 - VCML critical experiments.

The overall average and standard deviation of the sixty-three cases is 0.9948 ± 0.0044 . The average Monte Carlo error (statistical convergence) is ± 0.0012 for the sixty-three cases. This uncertainty component is statistically subtracted from the uncertainties, because it is previously included in the above standard deviation. The KENO-Va models are three dimensional, fully explicit representations (no homogenization) of the experimental geometry. Therefore, the uncertainty due to limitations of geometrical modeling is taken to be 0.0. The experiments modeled cover the range of fuel types, enrichments, neutron absorber configurations, neutron absorber ^{10}B loading and metal reflector effects, so there are no extrapolations necessary outside of the range of data, and the

uncertainty due to this is also taken to be 0.0. Based on the reported experimental error for the B&W cases, the reported error of the critical size number of rods for the PNL cases and the reported error for the critical height in the VCML cases, the experimental error is conservatively taken to be ± 0.001 . Criticality can then be represented as 1.000 ± 0.001 . This uncertainty component is statistically added to the sum of the other uncertainties because the bias is the difference between two random variates (i.e., criticality and code prediction, and the uncertainty in the difference between two random variates is the statistical sum (rms) of their individual uncertainties).

Thus, the bias or average difference between code prediction and critical is $\beta = 1 - 0.9948 = 0.0052$. The uncertainty in the bias, accounting for the statistical convergence (Monte Carlo error) and the uncertainty in criticality, is $(0.0044^2 - 0.0012^2 + 0.0010^2)^{1/2} = 0.0043$. For 63 samples of criticality, the 95/95 one side tolerance factor is 2.012 [18]. This results in a 95/95 one-sided uncertainty in the bias of $\Delta\beta = 2.012 \times 0.0043 = 0.0087$. Equation 4 now becomes:

$$k_{\text{eff}} + \Delta k_s + 0.0052 + 0.0087 \leq 0.95 \quad (5)$$

where Δk_s becomes the uncertainty in k_s due to Monte Carlo error, mechanical and material tolerances, and geometric or material representations. If the nominal representation of the system is evaluated for k_s , then the mechanical and material perturbation can be evaluated independently and can be combined statistically as the root sum of squares. If the worst case mechanical and material tolerances are used in the calculations of k_s (e.g., the most reactive positioning of fuel or basket components and 75% of the specified minimum boron loading), then Δk_s becomes 0.0 and the Monte Carlo error, σ_{mc} , can be combined statistically, since it is independent, with the uncertainty in the bias as:

$$k_{\text{eff}} + 0.0052 + \sqrt{0.0087^2 + (2\sigma)^2} \leq 0.95 \quad (6)$$

6.5.1.4 Trends

Scatter plots of k_{eff} versus wt % ^{235}U , rod pitch, H/U volume ratio, average neutron group causing fission, ^{10}B loading for flux trap cases, and flux trap gap thickness are shown in Figures 6.5.1-2 through 6.5.1-7. Included in these scatter plots are linear regression lines with a corresponding correlation coefficient (R). This statistically indicates any trend, or lack thereof. In particular, the

correlation coefficient is a measure of the linear relationship between k_{eff} and a critical experiment parameter. If R is +1, a perfect linear relationship with a positive slope is indicated, and if R is -1, a perfect linear relationship with a negative slope is indicated. When R is 0, no linear relationship is indicated.

The largest correlation coefficient indicated in the plots is +0.3608 (k_{eff} versus enrichment) and the lowest is +0.0693 (k_{eff} versus ^{10}B loading in flux trap experiments). Based on the correlation coefficients, no statistically significant trends exist over the range of variables studied. Most importantly, no significant trends are shown with either flux trap gap spacing or ^{10}B loading. This is the major criticality control feature of both the directly loaded basket and the Yankee-MPC canistered basket for Yankee Class fuel.

6.5.1.5 Comparison of NAC Method to NUREG/CR-6361

NUREG/CR-6361, "Criticality Benchmark Guide for Light-Water-Reactor Fuel in Transportation and Storage Packages," provides a guide to LWR criticality benchmark calculations and the determination of bias and subcritical limits in critical safety evaluations. In Section 2 of NUREG/CR-6361, a series of LWR critical experiments are described in sufficient detail for independent modeling. In Section 3, the critical experiments are modeled, and the results (k_{eff} values) are presented. The method utilized in the NUREG is KENO-Va with the 44 group ENDF/B-V cross section library embedded in the SCALE 4.3 code package. Inputs are provided in NUREG Appendix A. In Section 4, a guide for the determination of bias and subcritical safety limits and statistical analysis of the trending in the bias are provided based on ANSI/ANS-8.17. Finally, in Section 5, guidelines for experiment selection and applicability are presented. In this section, the approach presented in Section 4 of the NUREG is described in detail and is compared to the NAC approach presented in Sections 6.5.1, 6.5.1.1 and 6.5.1.2.

NAC has performed an extensive LWR critical benchmarking effort as documented in Sections 6.5.1.1 and 6.5.1.2. The method used in NAC benchmarking/validation included the CSAS25 (KENO-Va) criticality analysis sequence, with the 27 group ENDF/B-IV library, contained in the SCALE 4.3 package. Trending in k_{eff} was evaluated for the following independent variables: wt % ^{235}U , rod pitch, H/U volume ratio, average neutron group causing fission, ^{10}B loading for flux trap cases, and flux trap gap thickness. No statistically significant trends were found, and a constant bias with associated uncertainty was determined for criticality evaluation.

Both the NUREG/CR-6361 and the NAC approach to criticality evaluation start with ANSI/ANS-8.17 [19] criticality safety criterion:

$$k_s \leq k_c - \Delta k_s - \Delta k_c - \Delta k_m \quad (1)$$

where:

k_s = calculated allowable maximum multiplication factor, k_{eff} , of the system being evaluated for all normal or credible abnormal conditions or events.

k_c = mean k_{eff} that results from a calculation of benchmark criticality experiments using a particular calculation method. If the calculated k_{eff} values for the criticality experiments exhibit a trend with an independent parameter, then k_c shall be determined by extrapolation based on best fit to calculated values. Criticality experiments used as benchmarks in computing k_c should have physical compositions, configurations, and nuclear characteristics (including reflectors) similar to those of the system being evaluated.

Δk_s = allowance for:

- a) statistical or convergence uncertainties, or both, in computation of k_s ,
- b) material and fabrication tolerances, and
- c) geometric or material representations used in computational method.

Δk_c = margin for uncertainty in k_c which includes allowance for:

- a) uncertainties in critical experiments,
- b) statistical or convergence uncertainties, or both, in computation of k_c ,
- c) uncertainties resulting from extrapolation of k_c outside range of experimental data, and
- d) uncertainties resulting from limitations in geometrical or material representations used in computational method.

Δk_m = arbitrary administrative margin to ensure subcriticality of k_s .

The various uncertainties are combined statistically if they are independent. Correlated uncertainties are combined by addition.

Equation 1 can be rewritten as:

$$k_s \leq 1 - \Delta k_m - \Delta k_s - (1 - k_c) - \Delta k_c \quad (2)$$

Noting that the definition of the bias is $\beta = 1 - k_c$, Equation 2 can be written as:

$$k_s + \Delta k_s \leq 1 - \Delta k_m - \beta - \Delta \beta \quad (3)$$

where $\Delta \beta = \Delta k_c$. Thus, the maximum allowable value for k_{eff} plus uncertainties in the system being analyzed must be below 1 minus an administrative margin (typically 0.05), the bias and the uncertainty in the bias. This can also be written as:

$$k_s + \Delta k_s \leq USL \quad (4)$$

where,

$$USL \equiv 1 - \Delta k_m - \beta - \Delta \beta \quad (5)$$

This is the Upper Safety Limit criterion as described in Section 4 of NUREG/CR-6361. Two methods are prescribed for the statistical determination of the USL: Confidence Band with Administrative Margin (USL-1) and Single Sided Uniform with Close Approach (USL-2). In the first method, $\Delta k_m = 0.05$ and a lower confidence band (usually 95%) is specified based on a linear regression of k_{eff} as a function of some system parameter. In the second method, the arbitrary administrative margin is set to zero and a uniform lower tolerance band is determined based on a linear regression. Thus, the second method provides a criticality safety margin that is generally less than 0.05. In cases where there are a limited number of data points, this method may indicate the need for a larger administrative margin. In both cases, all the significant system parameters need to be studied to determine the strongest correlation.

In the Section 6.5.1.1 and 6.5.1.2 analyses, the bias and uncertainties are applied directly to the estimate of the system k_{eff} . Noting that the NRC requires a 5% subcriticality margin ($\Delta k_m = 0.05$), Equation 3 can be rewritten applying the bias and uncertainty in the bias to the k_{eff} of the system being analyzed as:

$$k_s + \Delta k_s + \beta + \Delta \beta \leq 0.95 \quad (6)$$

In Equation 6, the method bias and all uncertainties are added to k_s . This is the maximum k_{eff} criterion defined in Section 6.5.1.2.

To this point, both the USL criterion and maximum k_{eff} criterion are equivalent. The effects of trending in the bias or the uncertainty in the bias can be directly incorporated into either equation 5 or equation 6. Trending is established by performing a regression analysis of k_{eff} as a function of the principle system variables such as: enrichment, rod pitch, H to U ratio, average group of fission, ^{10}B absorber loading and flux trap gap spacing. Usually, simple linear regression is performed, and the line with the greatest correlation is used to functionalize β . This is the approach recommended in NUREG/CR-6361. However, if no strong correlation can be determined, then a constant bias adjustment can be made. This is typically done with a one-side tolerance factor that guarantees 95% confidence in the uncertainty in the bias. This is the approach taken in the criticality analysis for the NAC-STC directly loaded and Yankee-MPC canistered fuel configurations.

Both NUREG/CR-6361 and the NAC methodology performed regression analysis on key system parameters. For all the major system parameters, the NAC methodology found no strong correlation. This is based on the observation that the correlation coefficients are all much less than ± 1 . Thus, a constant bias with a 95/95 confidence factor is applied to the system k_{eff} . The NAC methodology's statistical analysis of the k_{eff} results produced a bias of 0.0052 and a 95/95 uncertainty of 0.0087. Adding the two together and subtracting from 0.95, yields an effective constant USL of 0.9361.

To assure compliance with NUREG/CR-6361, an upper safety limit is generated using USLSTATS and is compared to the constant NAC methodology bias and bias uncertainty used in Section 6.5.1.2.

To evaluate the relative importance of the trend analysis to the upper safety limits, correlation coefficients are required for all independent parameters. Table 6.5.1.2 contains the correlation coefficients, R , for each linear fit of k_{eff} versus experimental parameter (data is extracted from Figure 6.5.1.2 through Figure 6.5.1.7 by taking the square root of the listed R^2 value). Based on the highest correlation coefficient and the method presented in NUREG/CR-6361, a USL will be established based on the variation of k_{eff} with enrichment. Note that even the enrichment function shows a low statistical correlation coefficient (an $|R|$ equal or near 1 would indicate a good fit). The output generated by USLSTATS is shown in Figure 6.5.1.8.

The NAC methodology applied USL of 0.9361 bounds the calculated upper safety limits for all enrichment values above 3 wt % ^{235}U . Since the maximum reactivities in the NAC-STC cask systems are calculated at enrichments well above this level, the existing bias bounds the NUREG calculated USL. The parameters of the most reactive fuel element analysis are listed in Table 6.5.1-3.

Figure 6.5.1-1 KENO-Va Validation - 27 Group Library Results Frequency Distribution of K_{eff} Values

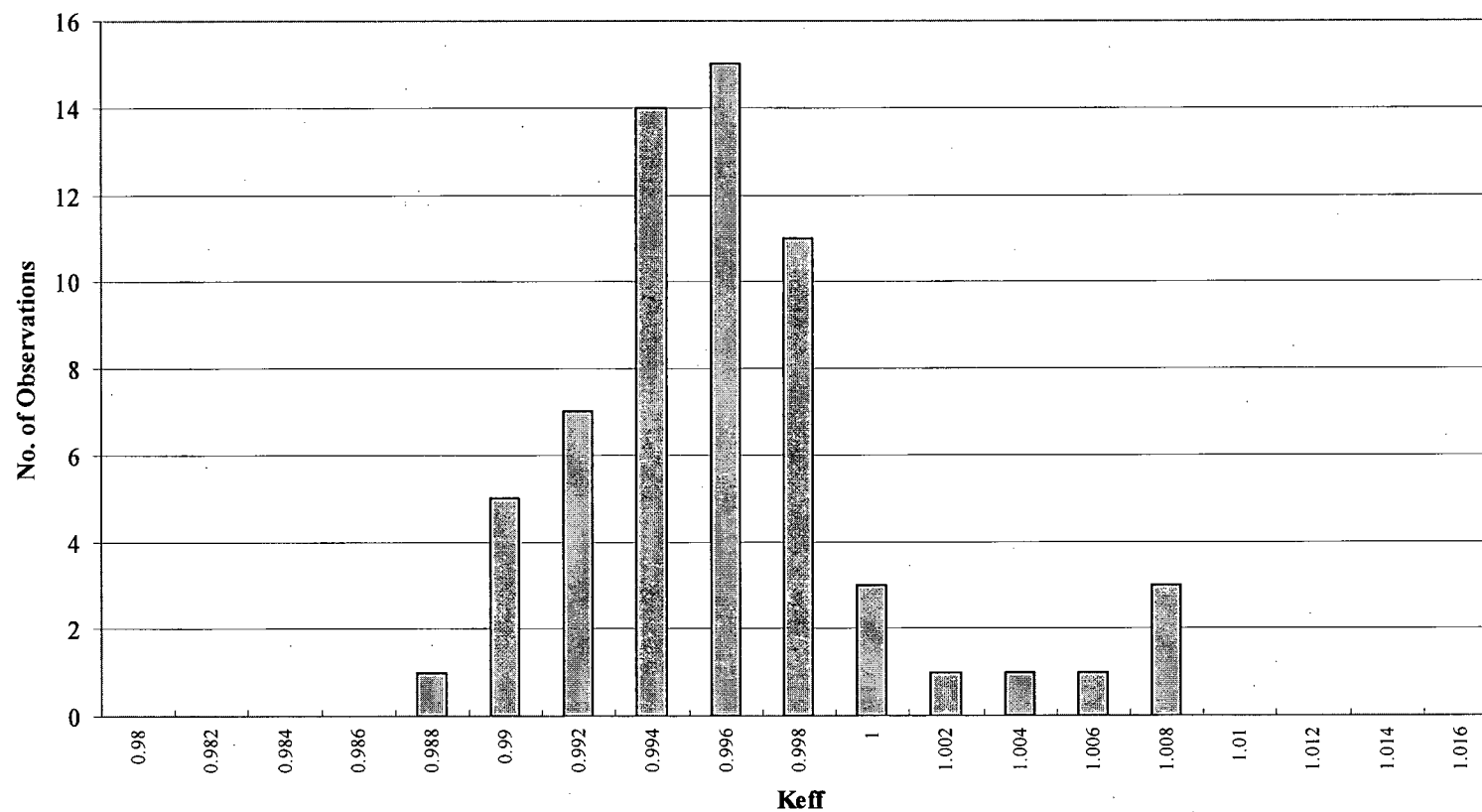


Figure 6.5.1-2 KENO-Va Validation -27 Group Library K_{eff} versus Enrichment

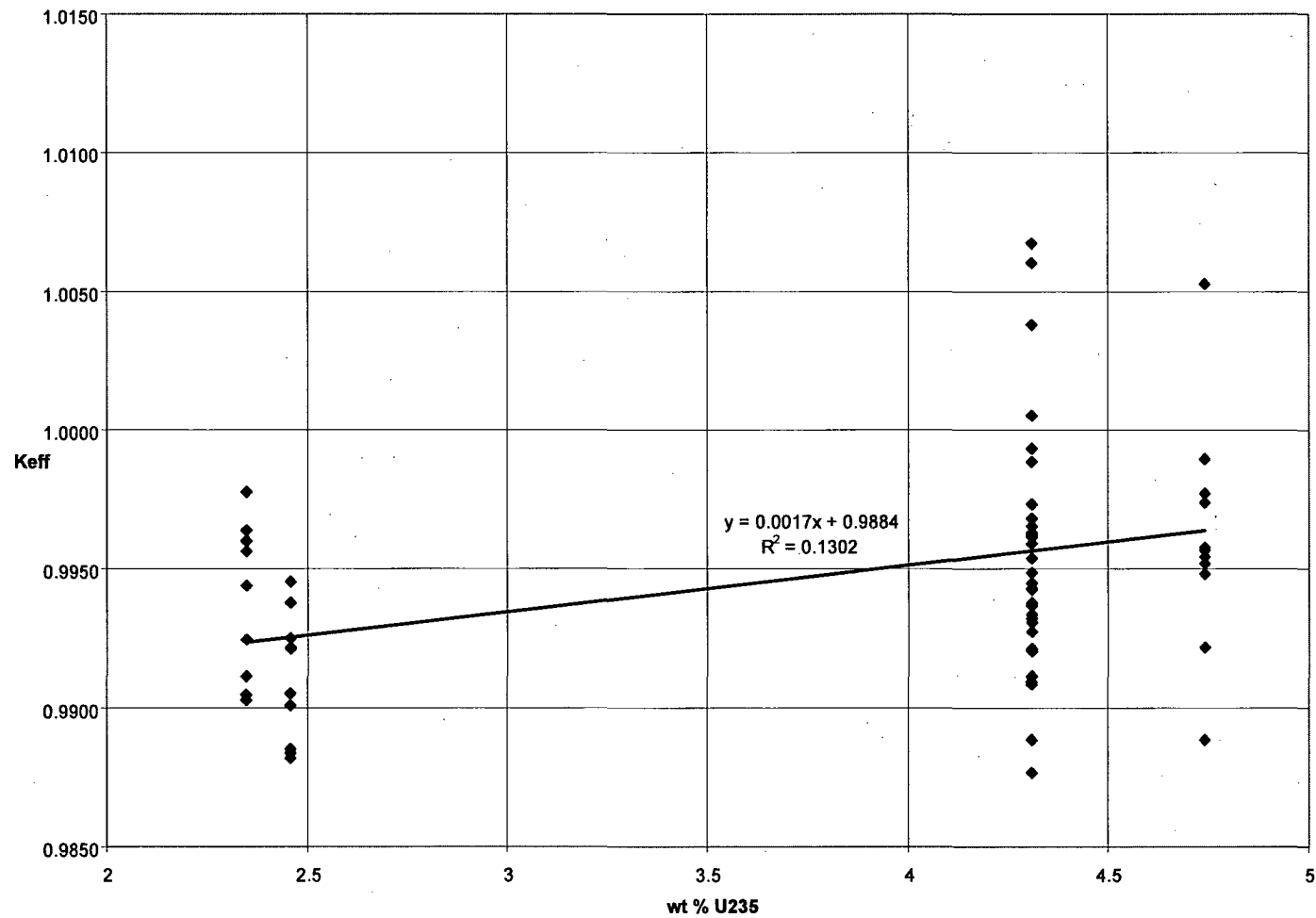


Figure 6.5.1-3 KENO-Va Validation - 27 Group Library K_{eff} versus Rod Pitch

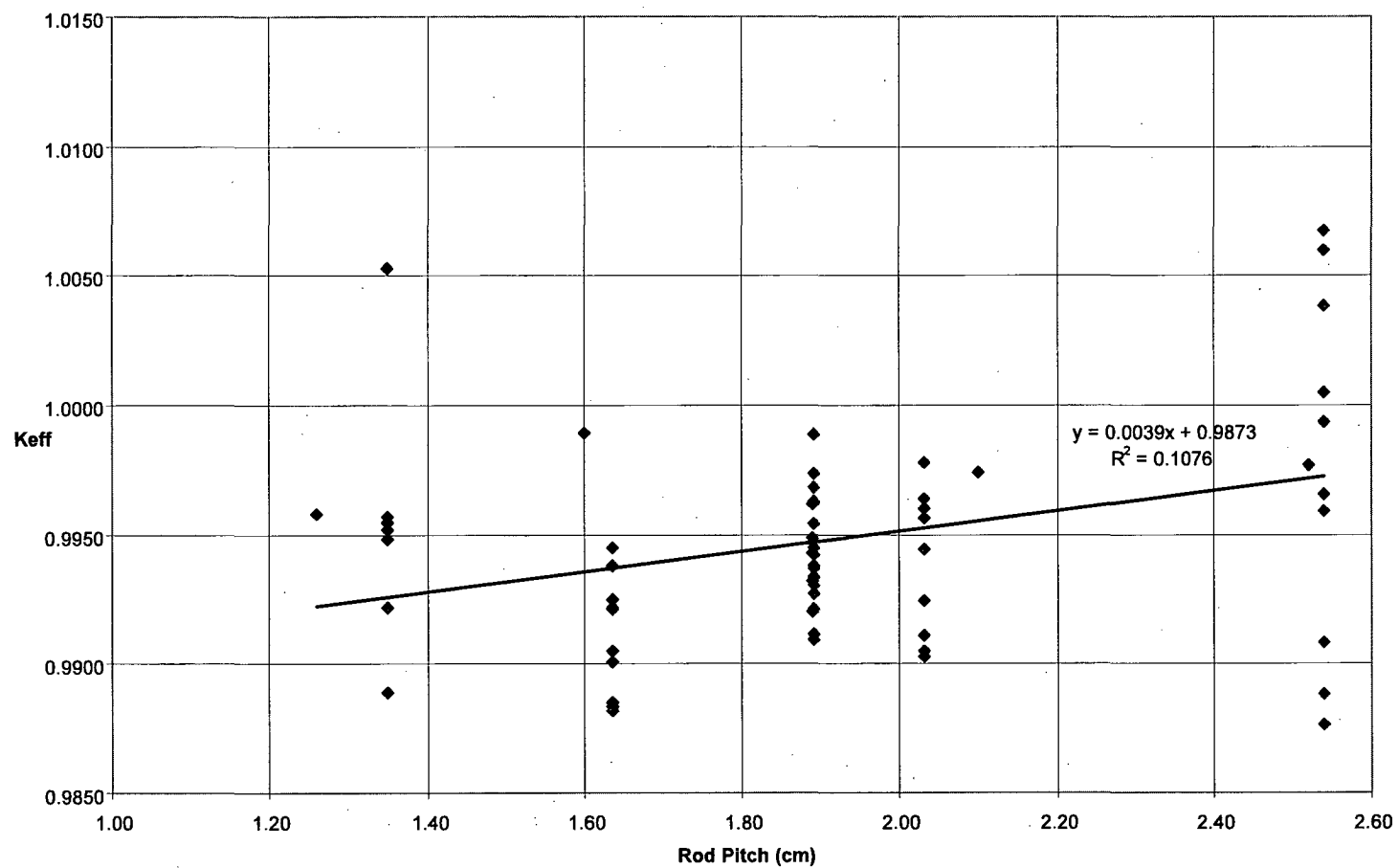


Figure 6.5.1-4 KENO-Va Validation -27 Group Library K_{eff} versus H/U Volume Ratio

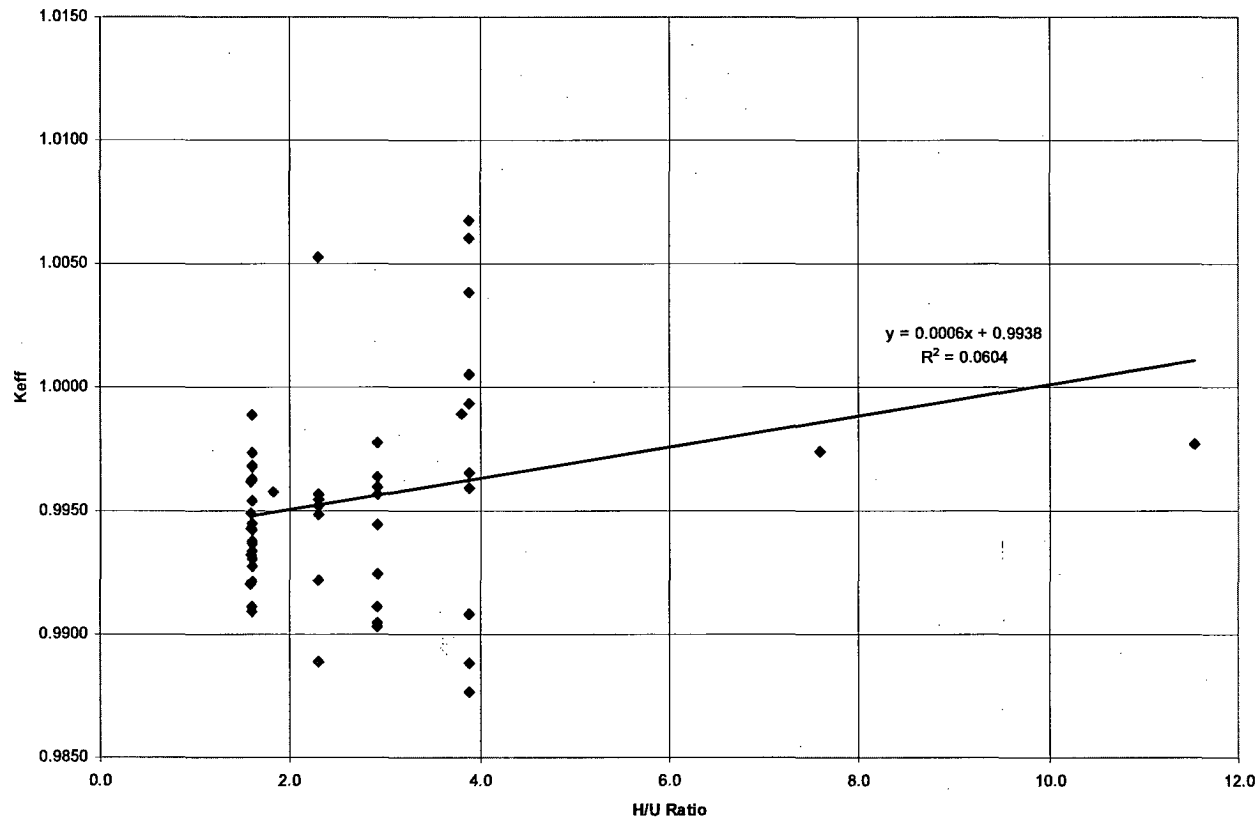


Figure 6.5.1-5 KENO-Va Validation -27 Group Library K_{eff} versus Average Group of Fission

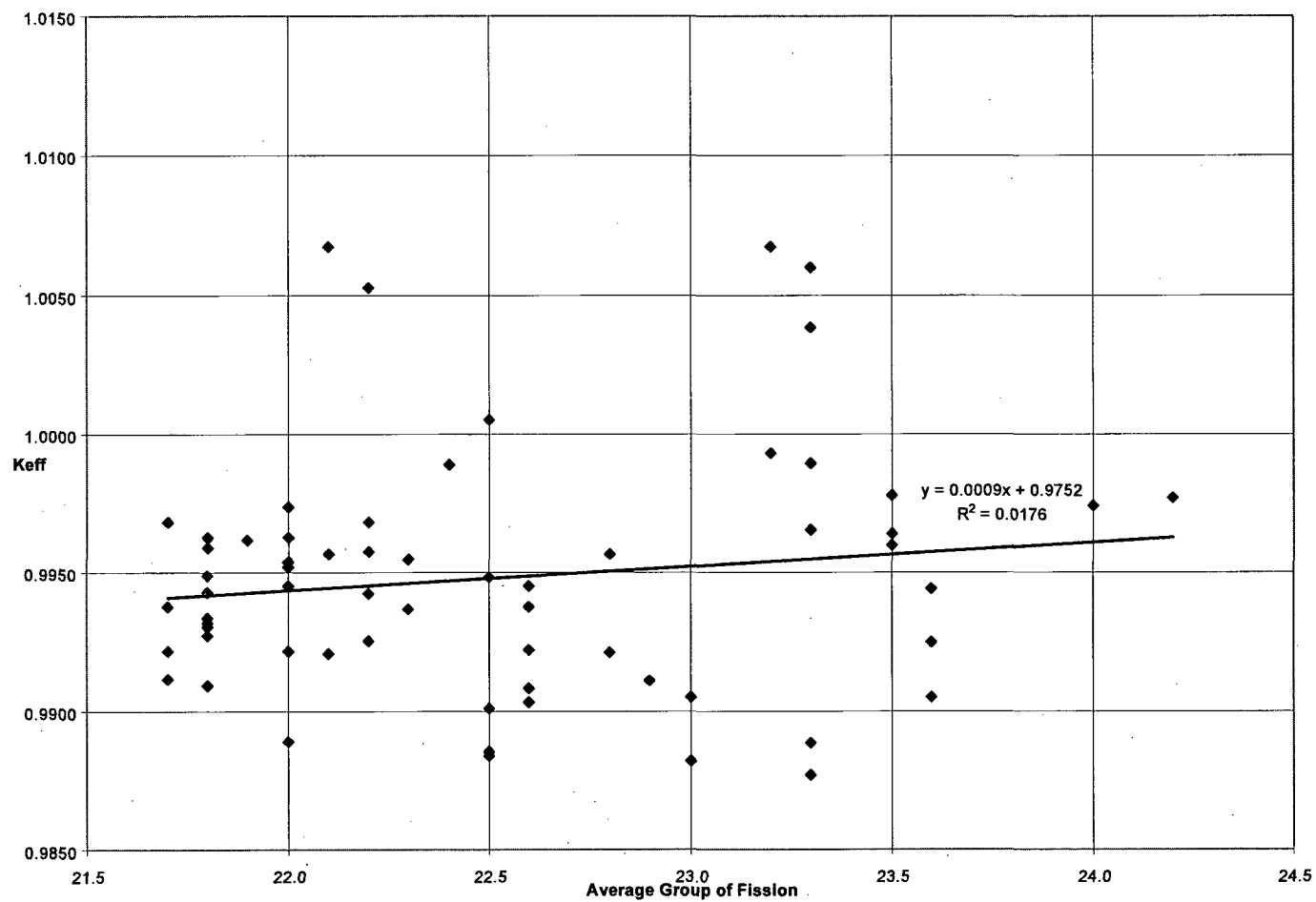


Figure 6.5.1-6 KENO-Va Validation - 27 Group Library K_{eff} versus ^{10}B Loading For Flux Trap Criticals

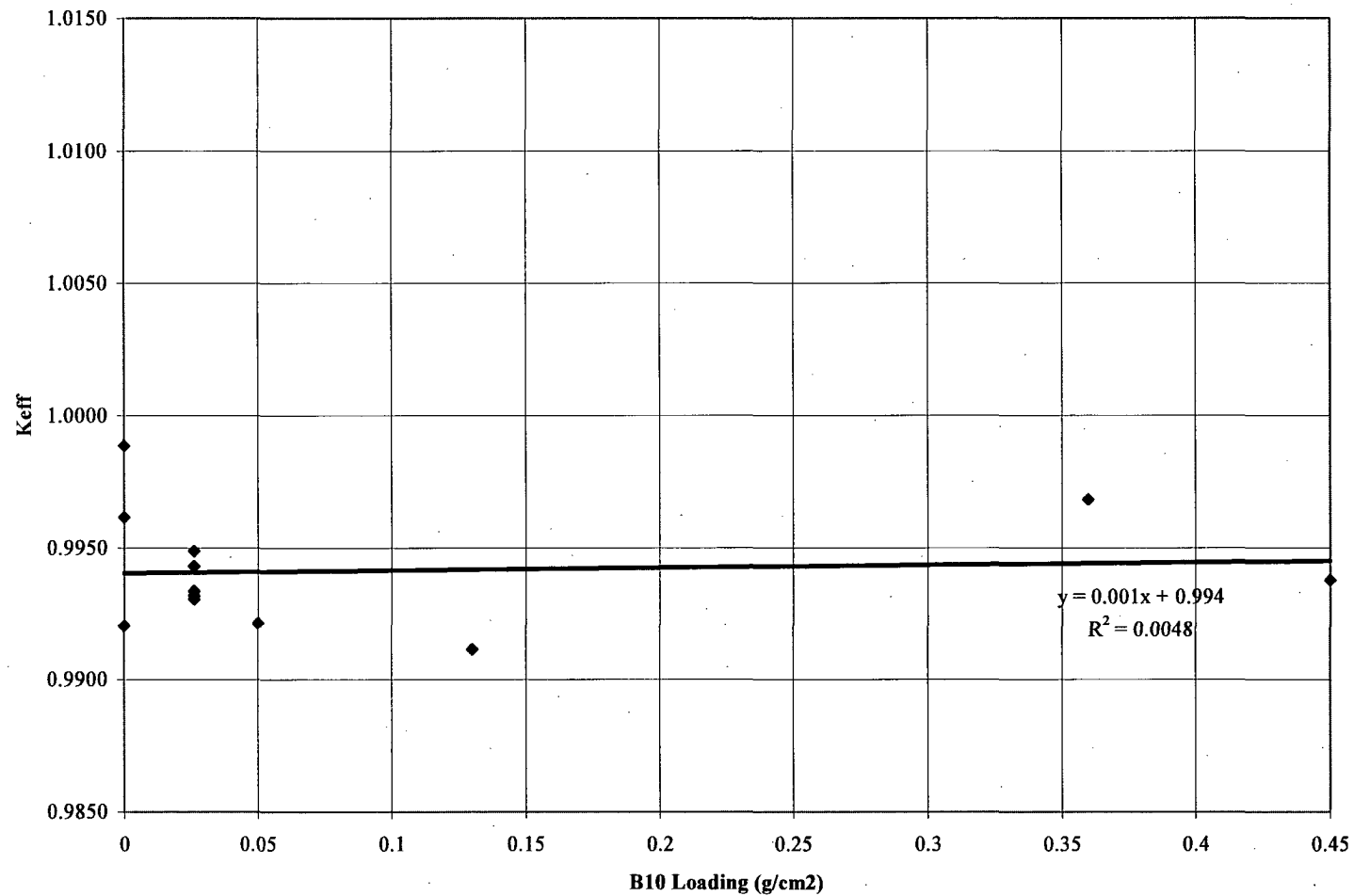


Figure 6.5.1-7 KENO-Va Validation -27 Group Library Results K_{eff} versus Flux Trap Critical Gap Thickness

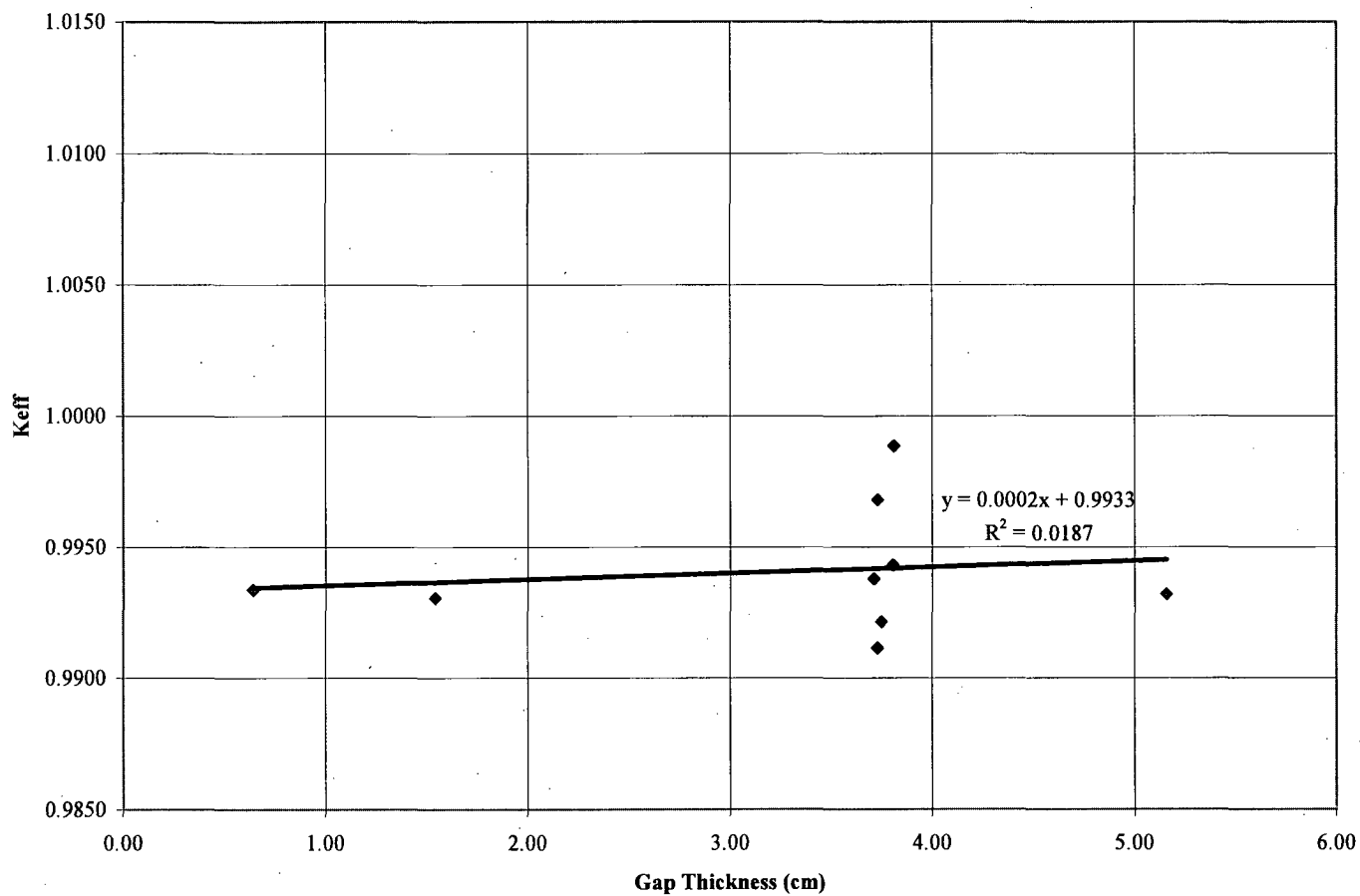


Figure 6.5.1-8 USLSTATS Output for Fuel Enrichment Study

uslstats: a utility to calculate upper subcritical
limits for criticality safety applications

Version 1.3.4, February 12, 1998
Oak Ridge National Laboratory

Input to statistical treatment from file:EN_KEFF.TXT

Title: 63 LWR CRITICAL EXPERIMENT KEFF VS ENRICHMENT

Proportion of the population = .995
Confidence of fit = .950
Confidence on proportion = .950
Number of observations = 63
Minimum value of closed band = 0.00
Maximum value of closed band = 0.00
Administrative margin = 0.05

independent variable - x	dependent variable - y	deviation in y	independent variable - x	dependent variable - y	deviation in y
2.35000E+00	9.96400E-01	1.00000E-03	4.31000E+00	9.96500E-01	1.10000E-03
2.35000E+00	9.94400E-01	1.00000E-03	4.31000E+00	1.00680E+00	2.10000E-03
2.35000E+00	9.90500E-01	1.00000E-03	4.31000E+00	1.00380E+00	1.20000E-03
2.35000E+00	9.96000E-01	1.10000E-03	4.31000E+00	9.88900E-01	1.10000E-03
2.35000E+00	9.97800E-01	1.00000E-03	4.31000E+00	9.95900E-01	1.10000E-03
2.35000E+00	9.92500E-01	1.00000E-03	4.31000E+00	1.00670E+00	1.00000E-03
2.35000E+00	9.90300E-01	9.00000E-04	4.31000E+00	1.00050E+00	1.10000E-03
2.35000E+00	9.95700E-01	1.00000E-03	4.31000E+00	9.90800E-01	1.10000E-03
2.35000E+00	9.91100E-01	1.00000E-03	4.31000E+00	9.98900E-01	1.20000E-03
2.46000E+00	9.92100E-01	1.10000E-03	4.31000E+00	9.92100E-01	1.20000E-03
2.46000E+00	9.92500E-01	9.00000E-04	4.31000E+00	9.91100E-01	1.20000E-03
2.46000E+00	9.93800E-01	9.00000E-04	4.31000E+00	9.96800E-01	1.30000E-03
2.46000E+00	9.90500E-01	1.00000E-03	4.31000E+00	9.93800E-01	1.20000E-03
2.46000E+00	9.88200E-01	1.00000E-03	4.31000E+00	9.93400E-01	1.00000E-03
2.46000E+00	9.94500E-01	1.00000E-03	4.31000E+00	9.93100E-01	1.00000E-03
2.46000E+00	9.92200E-01	1.00000E-03	4.31000E+00	9.94300E-01	1.00000E-03
2.46000E+00	9.88500E-01	1.00000E-03	4.31000E+00	9.93200E-01	1.00000E-03
2.46000E+00	9.88400E-01	1.00000E-03	4.31000E+00	9.94900E-01	1.00000E-03
2.46000E+00	9.90100E-01	9.00000E-04	4.31000E+00	9.92000E-01	1.00000E-03
4.31000E+00	9.95400E-01	1.40000E-03	4.31000E+00	9.96200E-01	1.00000E-03
4.31000E+00	9.94500E-01	1.30000E-03	4.74000E+00	9.92200E-01	1.30000E-03
4.31000E+00	9.97400E-01	1.30000E-03	4.74000E+00	9.88900E-01	1.30000E-03
4.31000E+00	9.96300E-01	1.30000E-03	4.74000E+00	9.95700E-01	1.30000E-03
4.31000E+00	9.92700E-01	1.20000E-03	4.74000E+00	1.00530E+00	1.10000E-03
4.31000E+00	9.90900E-01	1.20000E-03	4.74000E+00	9.95500E-01	1.20000E-03
4.31000E+00	9.96200E-01	1.20000E-03	4.74000E+00	9.94800E-01	1.30000E-03
4.31000E+00	9.93700E-01	1.30000E-03	4.74000E+00	9.95800E-01	1.20000E-03
4.31000E+00	9.94200E-01	1.20000E-03	4.74000E+00	9.95200E-01	1.20000E-03
4.31000E+00	9.96800E-01	1.20000E-03	4.74000E+00	9.98900E-01	1.30000E-03
4.31000E+00	9.87700E-01	2.30000E-03	4.74000E+00	9.97400E-01	1.20000E-03
4.31000E+00	9.99300E-01	1.20000E-03	4.74000E+00	9.97700E-01	1.10000E-03
4.31000E+00	1.00600E+00	2.20000E-03			

chi = 2.1587 (upper bound = 9.49). The data tests normal.

Output from statistical treatment

Figure 6.5.1-8 USLSTATS Output for Fuel Enrichment Study (Continued)

63 LWR CRITICAL EXPERIMENT KEFF VS ENRICHMENT

Number of data points (n)	63
Linear regression, k(X)	0.9884 + (1.6748E-03)*X
Confidence on fit (1-gamma) [input]	95.0%
Confidence on proportion (alpha) [input]	95.0%
Proportion of population falling above lower tolerance interval (rho) [input]	99.5%
Minimum value of X	2.3500
Maximum value of X	4.7400
Average value of X	3.81143
Average value of k	0.99482
Minimum value of k	0.98770
Variance of fit, s(k,X)^2	1.6973E-05
Within variance, s(w)^2	1.4306E-06
Pooled variance, s(p)^2	1.8404E-05
Pooled std. deviation, s(p)	4.2900E-03
C(alpha,rho)*s(p)	1.5488E-02
student-t @ (n-2,1-gamma)	1.67078E+00
Confidence band width, W	7.3606E-03
Minimum margin of subcriticality, C*s(p)-W	8.1273E-03

Upper subcritical limits: (2.35000 <= X <= 4.74000)

USL Method 1 (Confidence Band with
Administrative Margin) USL1 = 0.9311 + (1.6748E-03)*X

USL Method 2 (Single-Sided Uniform
Width Closed Interval Approach) USL2 = 0.9729 + (1.6748E-03)*X

USLs Evaluated Over Range of Parameter X:

X:	2.35	2.69	3.03	3.37	3.72	4.06	4.40	4.74

USL-1:	0.9350	0.9356	0.9362	0.9367	0.9373	0.9379	0.9384	0.9390
USL-2:	0.9769	0.9775	0.9780	0.9786	0.9792	0.9797	0.9803	0.9809

Thus spake USLSTATS
Finis.

Table 6.5.1-1 KENO-Va and 27 Group Library Validation Statistics

Criticals	Configuration	wt % ²³⁵ U	Pitch (cm)	Clad OD (cm)	Pellet OD (cm)	H/U	Sol. B (ppm)	Poison	g ¹⁰ B/cm ²	Gap(cm)	Gap Den.	Average FissionGroup	K _{eff}	σ
Set 1														
B&W-I	Cylindrical	2.46	1.636	1.206	1.03	1.6	0	na	na	0		22.8	0.9921	0.0011
B&W-II	3X3-14X14	2.46	1.636	1.206	1.03	1.6	1037	na	na	0		22.2	0.9925	0.0009
B&W-III	3X3-14X14	2.46	1.636	1.206	1.03	1.6	764	na	na	1.636		22.6	0.9938	0.0009
B&W-IX	3X3-14X14	2.46	1.636	1.206	1.03	1.6	0	na	na	6.543		23	0.9905	0.0010
B&W-X	3X3-14X14	2.46	1.636	1.206	1.03	1.6	143	na	na	4.907		23	0.9882	0.0010
B&W-XI	3X3-14X14	2.46	1.636	1.206	1.03	1.6	514	Steel	0	1.636		22.6	0.9945	0.0010
B&W-XIII	3X3-14X14	2.46	1.636	1.206	1.03	1.6	15	B-Al	0.0052	1.636		22.6	0.9922	0.0010
B&W-XIV	3X3-14X14	2.46	1.636	1.206	1.03	1.6	92	B-Al	0.0040	1.636		22.5	0.9885	0.0010
B&W-XVII	3X3-14X14	2.46	1.636	1.206	1.03	1.6	487	B-Al	0.0008	1.636		22.5	0.9884	0.0010
B&W-XIX	3X3-14X14	2.46	1.636	1.206	1.03	1.6	634	B-Al	0.0003	1.636		22.5	0.9901	0.0009
												Average	0.9911	0.0023
Set 2														
PNL-043	17X13 Lattice	4.31	1.892	1.415	1.265	1.6	0	na	na	na	na	22.0	0.9954	0.0014
PNL-044	16X14 Lattice	4.31	1.892	1.415	1.265	1.6	0	na	na	na	na	22.0	0.9945	0.0013
PNL-045	14X16 Lattice	4.31	1.892	1.415	1.265	1.6	0	na	na	na	na	22.0	0.9974	0.0013
PNL-046	12x19 Lattice	4.31	1.892	1.415	1.265	1.6	0	na	na	na	na	22.0	0.9963	0.0013
PNL-087	4 11X14 Arrays	4.31	1.892	1.415	1.265	1.6	0	BORAL	0.066	2.83		21.8	0.9927	0.0012
PNL-079	4 11X14 Arrays	4.31	1.892	1.415	1.265	1.6	0	BORAL	0.030	2.83		21.8	0.9909	0.0012
PNL-093	4 11X14 Arrays	4.31	1.892	1.415	1.265	1.6	0	BORAL	0.026	2.83		21.8	0.9962	0.0012
PNL-115	4 9X12 Arrays	4.31	1.892	1.415	1.265	1.6	0	Aluminum	0	2.83		22.3	0.9937	0.0013
PNL-064	4 9X12 Arrays	4.31	1.892	1.415	1.265	1.6	0	Steel (.302)	0	2.83		22.2	0.9942	0.0012
PNL-071	4 9X12 Arrays	4.31	1.892	1.415	1.265	1.6	0	Steel (.485)	0	2.83		22.2	0.9968	0.0012
												Average	0.9948	0.0020

Table 6.5.1-1 KENO-Va and 27 Group Library Validation Statistics (continued)

Criticals	Configuration	wt % ²³⁵ U	Pitch (cm)	Clad OD (cm)	Pellet OD (cm)	H/U	Sol. B (ppm)	Poison	g ¹⁰ B/cm ²	Gap(cm)	Gap Den.	Average Fission Group	K _{eff}	σ
Set 3										Cluster	Wall/Cluster			
PNL-STA	3X1 St Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	10.65	0.00	23.5	0.9964	0.0010
PNL-STB	3X1 St Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	11.20	1.32	23.6	0.9944	0.0010
PNL-STC	3X1 St Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	10.36	2.62	23.6	0.9905	0.0010
PNL-PBA	3X1 Pb Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	13.84	0.00	23.5	0.9960	0.0011
PNL-PBB	3X1 Pb Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	13.72	0.66	23.5	0.9978	0.0010
PNL_PBC	3X1 Pb Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	11.25	2.62	23.6	0.9925	0.0010
PNL-DUA	3X1 DU Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	11.83	0.00	22.6	0.9903	0.0009
PNL-DUB	3X1 DU Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	14.11	1.96	22.8	0.9957	0.0010
PNL-DUC	3X1 DU Refl.	2.35	2.032	1.27	1.1176	2.9	0	na	na	13.70	2.62	22.9	0.9911	0.0010
PNL-H20	3X1 H2O Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	8.24	inf	23.3	0.9877	0.0023
PNL-ST0	3X1 St Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	12.89	0	23.2	0.9993	0.0012
PNL-ST1	3X1 St Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	14.12	1.32	23.3	1.0060	0.0022
PNL-ST26	3X1 St Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	12.44	2.62	23.3	0.9965	0.0011
PNL-PB0	3X1 Pb Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	20.62	0	23.2	1.0068	0.0021
PNL-PB13	3X1 Pb Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	19.04	1.32	23.3	1.0038	0.0012
PNL-PB5	3X1 Pb Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	10.3	5.41	23.3	0.9889	0.0011
PNL-DU0	3X1 DU Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	15.38	0	21.8	0.9959	0.0011
PNL-DU13	3X1 DU Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	19.04	1.32	22.1	1.0067	0.0010
PNL-DU39	3X1 DU Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	18.05	3.91	22.5	1.0005	0.0011
PNL-DU54	3X1 DU Refl.	4.31	2.54	1.415	1.265	3.9	0	na	na	13.49	5.41	22.6	0.9908	0.0011
												Average	0.9964	0.0060

Table 6.5.1-1 KENO-Va and 27 Group Library Validation Statistics (continued)

Criticals	Configuration	wt % ²³⁵ U	Pitch (cm)	Clad OD (cm)	Pellet OD (cm)	H/U	Sol. B (ppm)	Poison	g ¹⁰ B/cm ²	Gap(cm)	Gap Den.	Average Fission Group	K _{eff}	σ
Set 4														
PNL-229	2x2 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	Aluminum	0	3.81	0.9982	22.4	0.9989	0.0012
PNL-230	2x2 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.05	3.75	0.9982	21.7	0.9921	0.0012
PNL-228	2x2 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.13	3.73	0.9982	21.7	0.9911	0.0012
PNL-214	2x2 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.36	3.73	0.9982	21.7	0.9968	0.0013
PNL-231	2x2 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.45	3.71	0.9982	21.7	0.9938	0.0012
PNL-127	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.026	0.64	0.9982	21.8	0.9934	0.0010
PNL-126	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.026	1.54	0.9982	21.8	0.9931	0.0010
PNL-123	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.026	3.80	0.9982	21.8	0.9943	0.0010
PNL-125	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.026	5.16	0.9982	21.8	0.9932	0.0010
PNL-124	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	BORAL	0.026	INF	0.9982	21.8	0.9949	0.0010
PNL-123-S	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	Steel	0	3.80	0.9982	22.1	0.9920	0.0010
PNL-124-S	2x1 Flux Trap	4.31	1.89	1.415	1.265	1.6	0	Steel	0	INF	0.9982	21.9	0.9962	0.0010
												Average	0.9941	0.0022
Set 5														
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	1.90	0	22.0	0.9922	0.0013
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	1.90	0.0323	22.0	0.9889	0.0013
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	1.90	0.2879	22.1	0.9957	0.0013
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	1.90	0.5540	22.2	1.0053	0.0011
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	2.50	0.9982	22.3	0.9955	0.0012
VCML	2x2 Water Gap	4.74	1.35	0.94	0.79	2.3	0	na	na	5.00	0.9982	22.5	0.9948	0.0013
VCML	Square Lattice	4.74	1.26	0.94	0.79	1.8	0	na	na	na	na	22.2	0.9958	0.0012
VCML	Square Lattice	4.74	1.35	0.94	0.79	2.3	0	na	na	na	na	22.0	0.9952	0.0012
VCML	Square Lattice	4.74	1.60	0.94	0.79	3.8	0	na	na	na	na	23.3	0.9989	0.0013
VCML	Square Lattice	4.74	2.10	0.94	0.79	7.6	0	na	na	na	na	24.0	0.9974	0.0012
VCML	Square Lattice	4.74	2.52	0.94	0.79	11.5	0	na	na	na	na	24.2	0.9977	0.0011
												Average	0.9961	0.0041

Table 6.5.1-2 Correlation Coefficient for Linear Curve-Fit of Critical Benchmarks

Correlation Studied	Correlation Coefficient (R)
k_{eff} versus enrichment	0.361
k_{eff} versus rod pitch	0.328
k_{eff} versus H/U volume ratio	0.246
k_{eff} versus ^{10}B loading	0.069
k_{eff} versus average group causing fission	0.133
k_{eff} versus flux gap thickness	0.137

Table 6.5.1-3 Most Reactive Configuration System Parameters

Parameters	Value
Enrichment (wt % ^{235}U) ¹	4.0
Rod pitch (cm)	1.1887
H/U volume ratio	1.52
^{10}B loading (g/cm^2)	0.01
Average group causing fission	21.6
Flux gap thickness (cm)	1.9 to 2.25

- 1.) Minor variations in the maximum enrichment due to fuel fabrication tolerances are considered in Section 6.4.2.1.

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6.5.2 MONK8a Validation in Accordance with NUREG/CR-6361

NUREG/CR-6361, "Criticality Benchmark Guide for Light-Water-Reactor Fuel in Transportation and Storage Packages" (NUREG), provides the guide to criticality benchmark calculations and the determination of bias and subcritical limits in critical safety evaluations. This section implements the ULSTATS method for the MONK8a application with JEF 2.2 point energy libraries.

Benchmarking of MONK8a has been performed by AEA Technologies. Critical benchmarks relevant to light water reactor spent fuel evaluations were extracted from the total benchmark set and listed in Table 6.5.2-1. Trending in k_{eff} was evaluated for the independent variables of enrichment, rod pitch, fuel pellet diameter, fuel rod diameter, H/U ratio, average neutron group causing fission, ^{10}B loading for flux trap cases, and flux trap gap thickness. Plots depicting the trend in k_{eff} as a function of each independent parameter are provided in Figures 6.5.2-1 through 6.5.2-8. The minimum and maximum values of each parameter define the area of applicability of the validation as summarized in Table 6.5.2-2.

To evaluate the relative importance of the trend analysis to the upper safety limit (USL), correlation coefficients are required for all independent parameters. Table 6.5.2-3 contains the correlation coefficient, R , for each linear fit of k_{eff} versus the experimental parameter (data is extracted from Figures 6.5.2-1 through 6.5.2-8 by taking the square root of the R^2 value). The USL is established based on the highest correlation coefficient, which occurs for the variation of k_{eff} with flux trap thickness, and the method presented in NUREG/CR-6361. Note that even the flux trap function shows a low statistical correlation coefficient (an $|R|$ equal or near 1 would indicate a good fit). The output generated by USLSTATS is shown in Figure 6.5.2-9.

The NAC applied USL is 0.9425, which bounds the calculated upper safety limits for the typical flux trap spacing found in multi-purpose casks. As demonstrated in Table 6.5.2-2, the parameters of the most reactive Connecticut Yankee fuel fall within the area of applicability of this validation.

Figure 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics – k_{eff} versus Fuel Enrichment

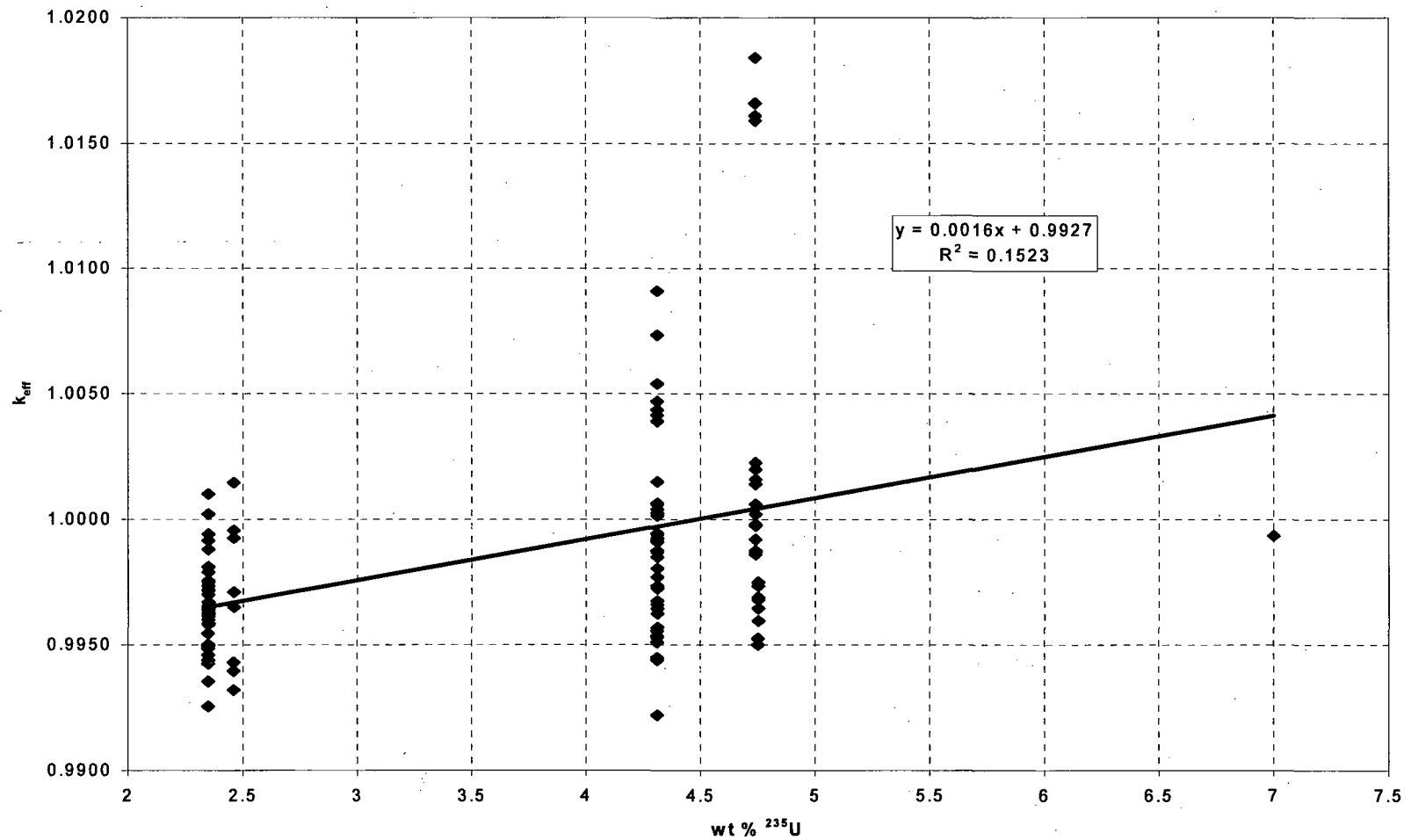


Figure 6.5.2-2 MONK8a – JEF 2.2 Library – k_{eff} versus Rod Pitch

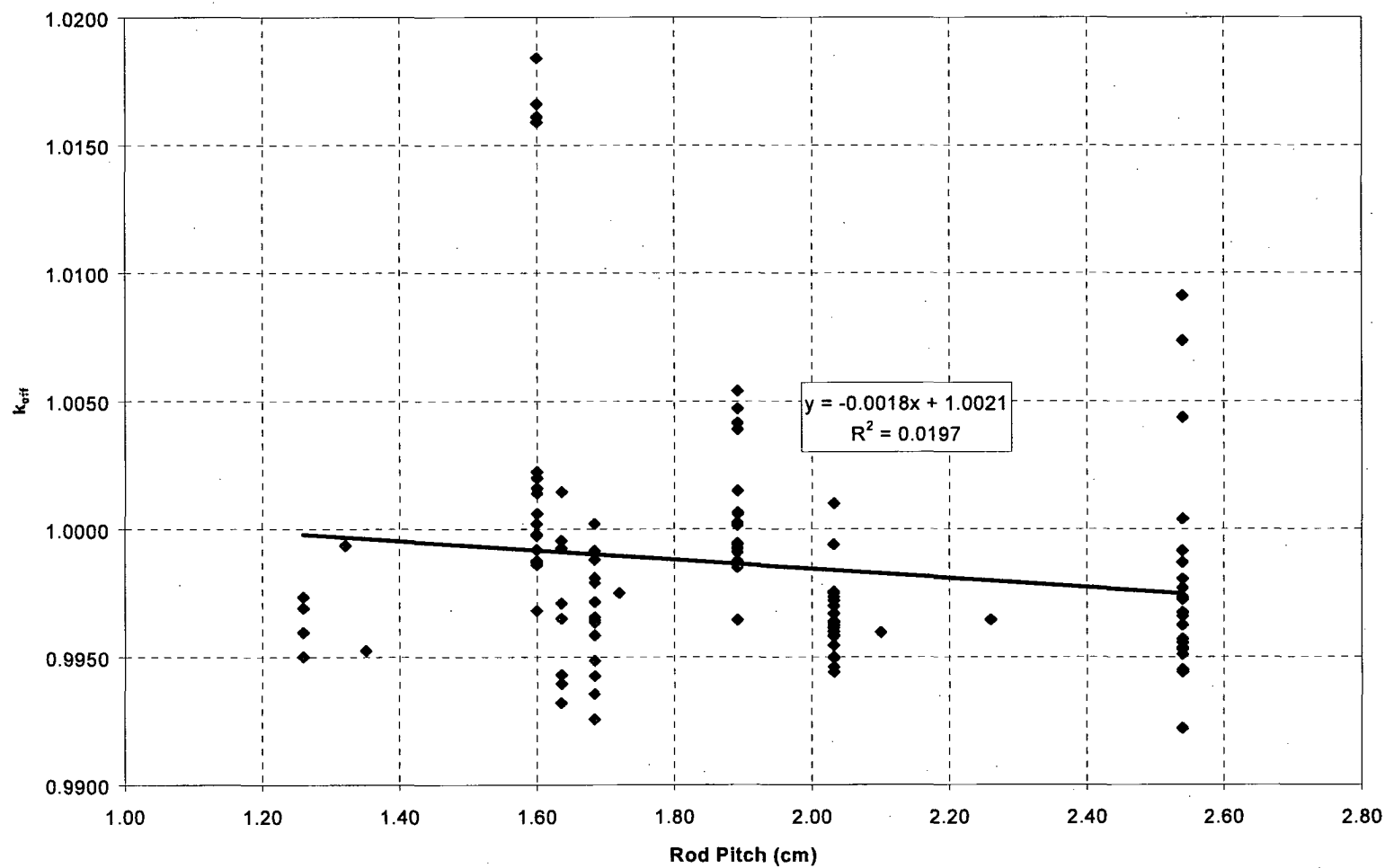


Figure 6.5.2-3 MONK8a – JEF 2.2 Library - k_{eff} versus H/U (fissile) Atom Ratio

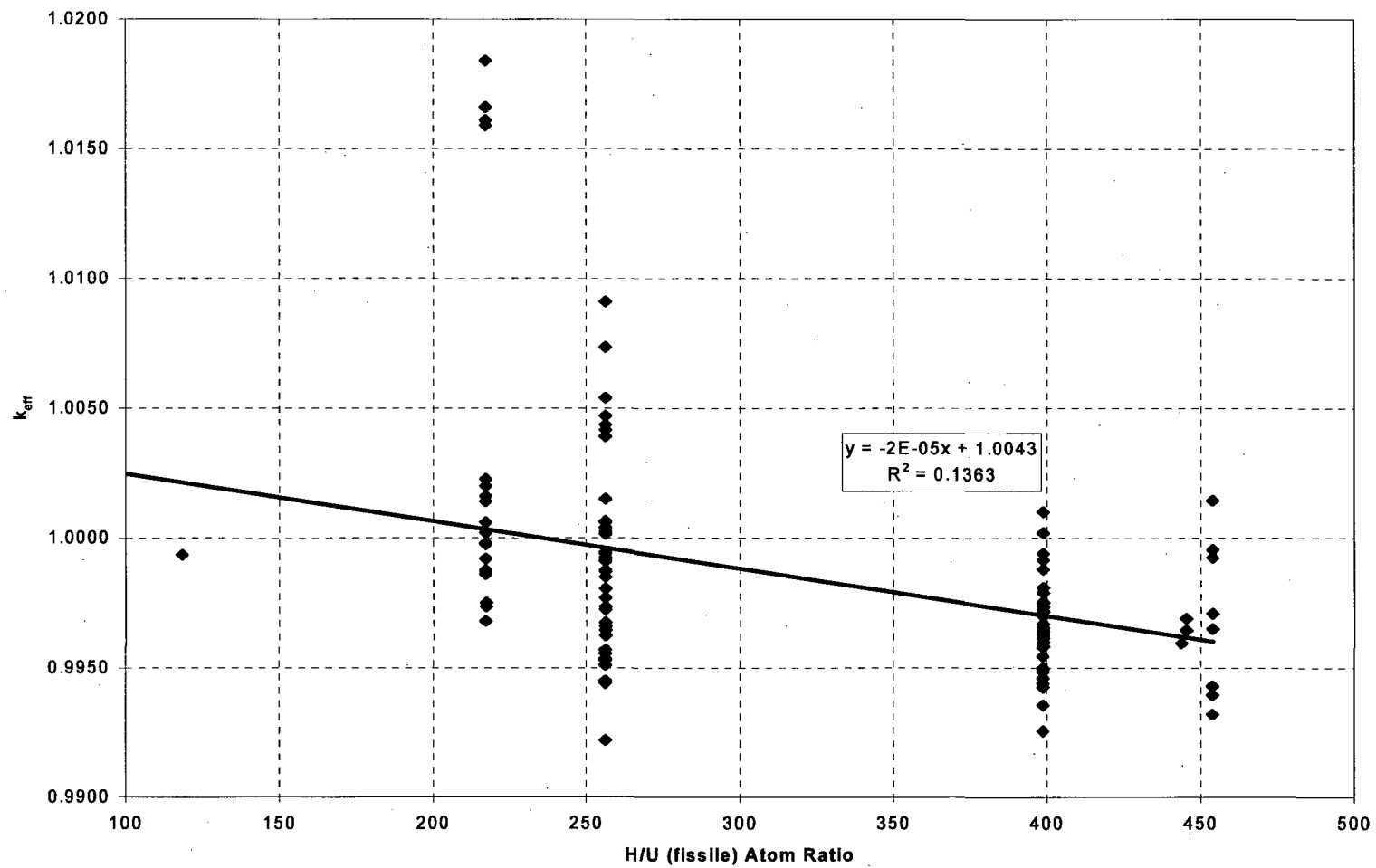


Figure 6.5.2-4 MONK8a – JEF 2.2 Library - k_{eff} versus ^{10}B Loading

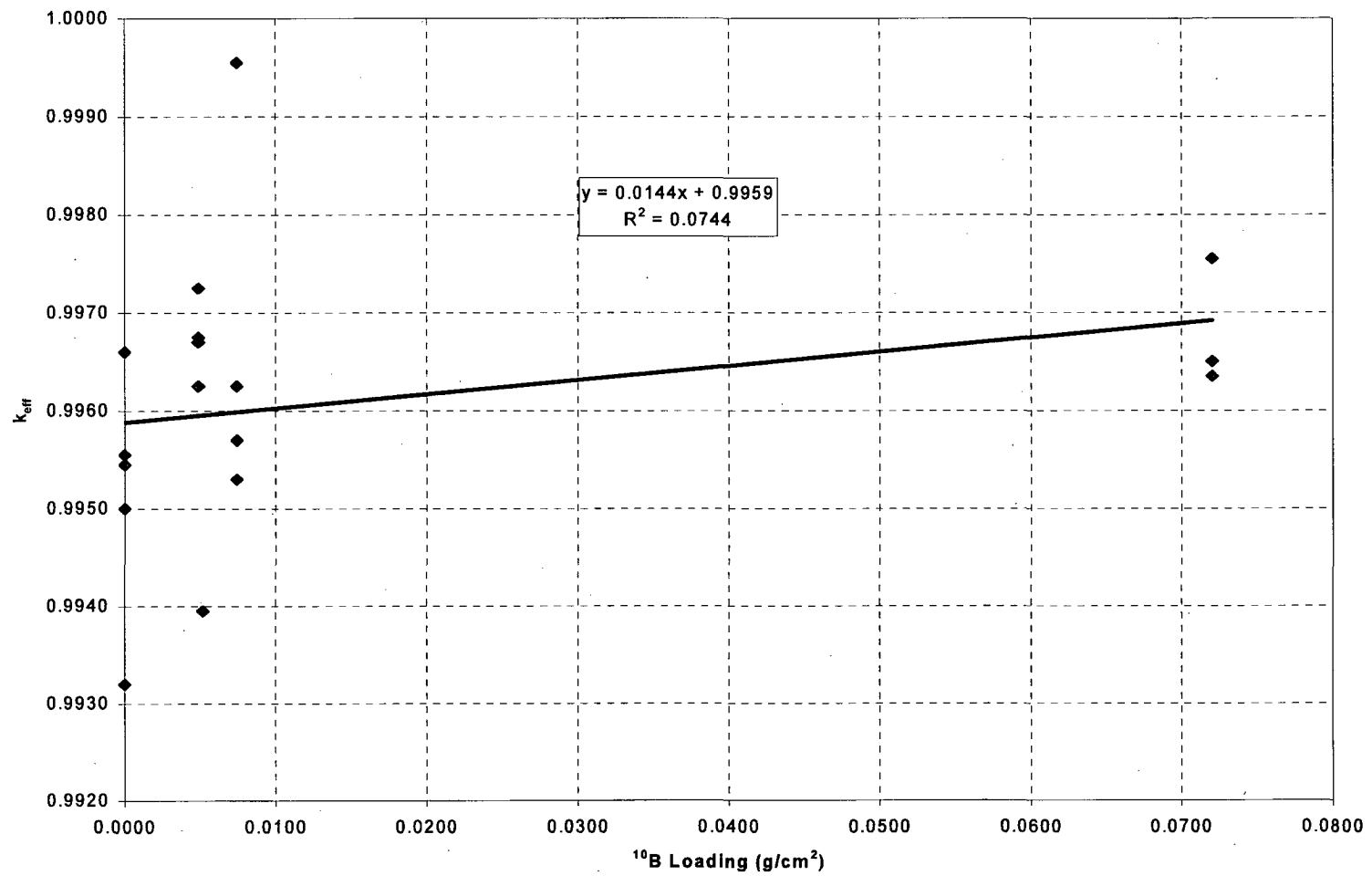


Figure 6.5.2-5 MONK8a – JEF 2.2 Library - k_{eff} versus Mean Neutron Log(e) Causing Fission

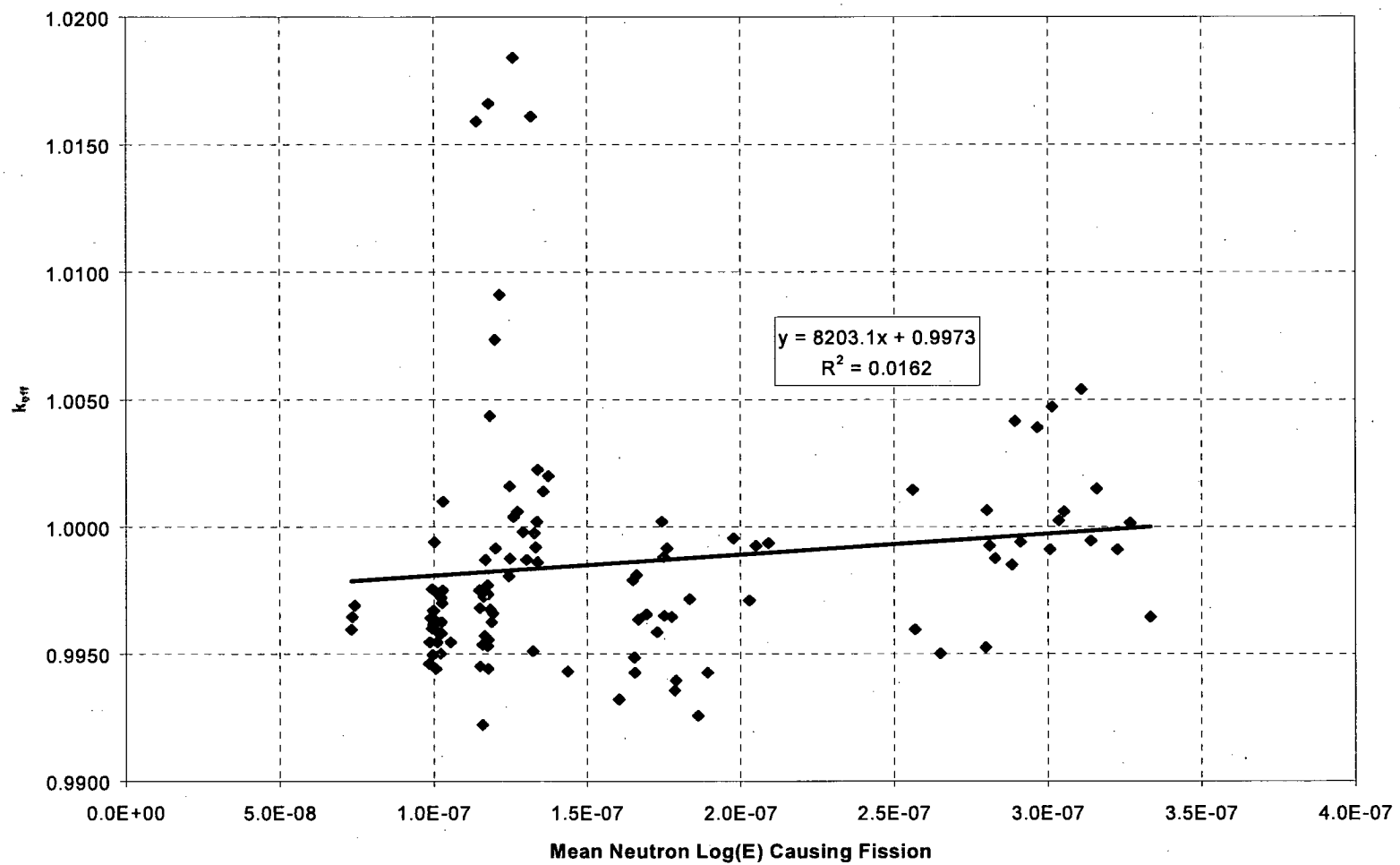


Figure 6.5.2-6 MONK8a – JEF 2.2 Library - k_{eff} versus Cluster Gap Thickness

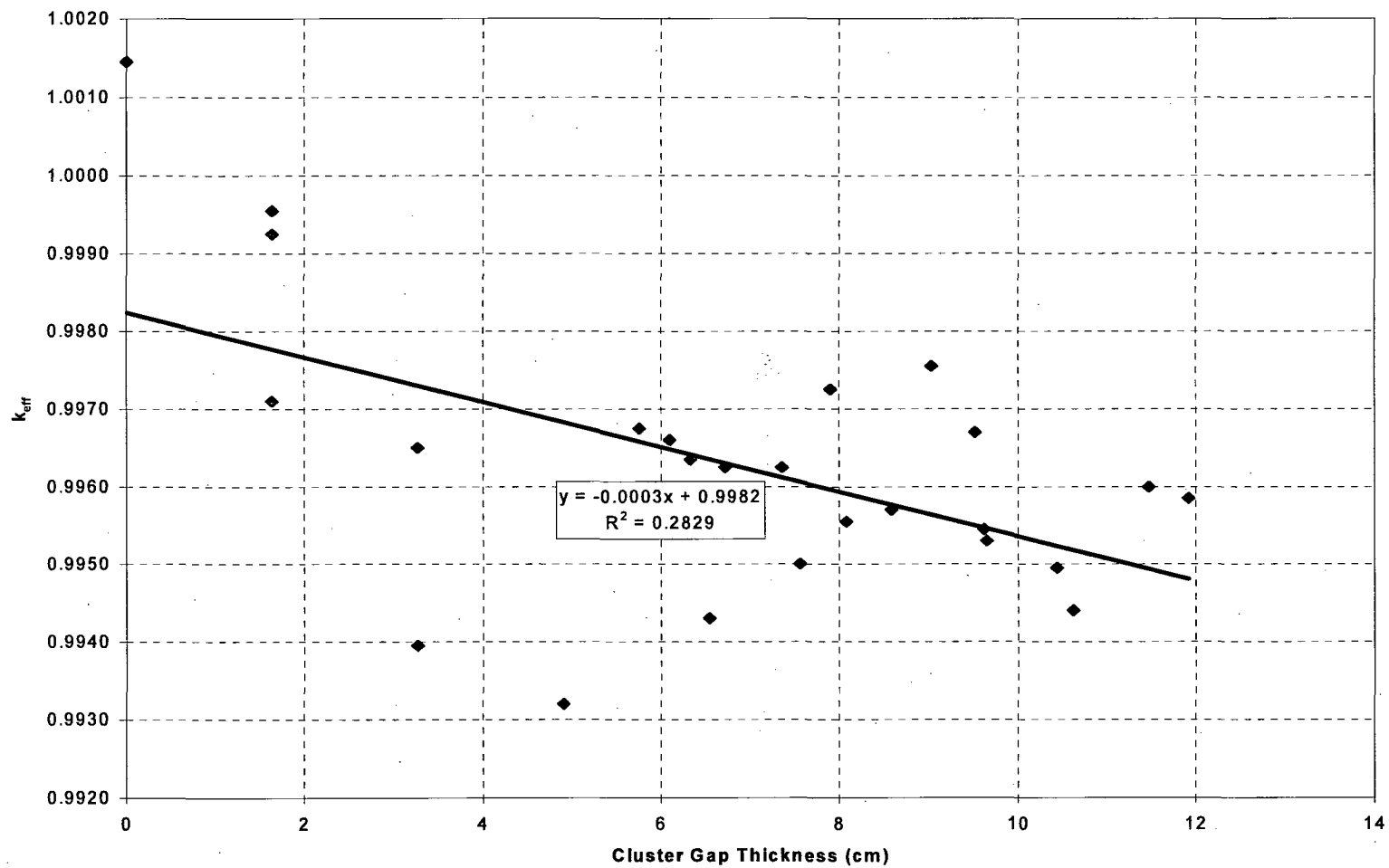


Figure 6.5.2-7 MONK8a – JEF 2.2 Library - k_{eff} versus Fuel Pellet Outside Diameter

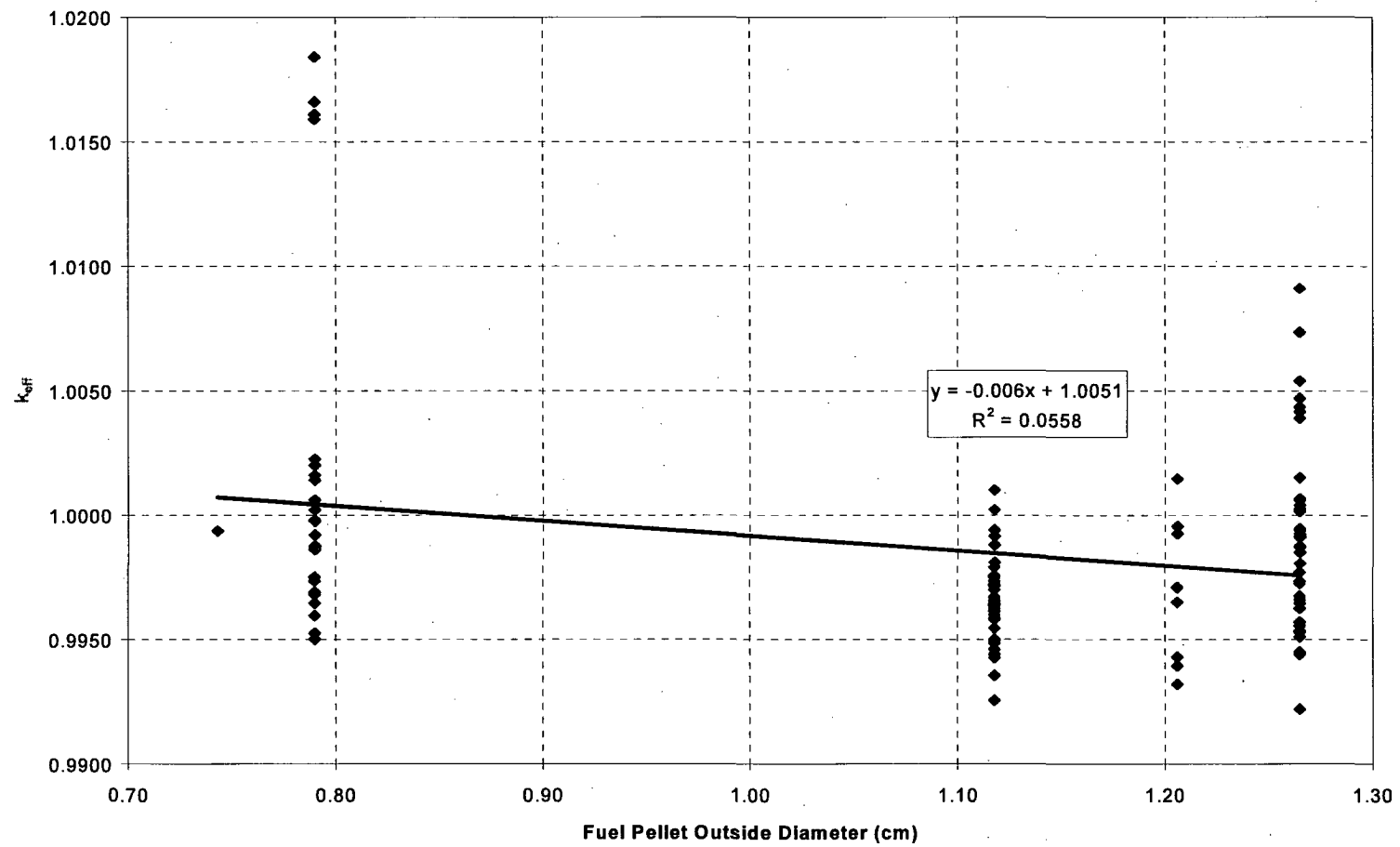


Figure 6.5.2-8 MONK8a – JEF 2.2 Library - k_{eff} versus Fuel Rod Outside Diameter

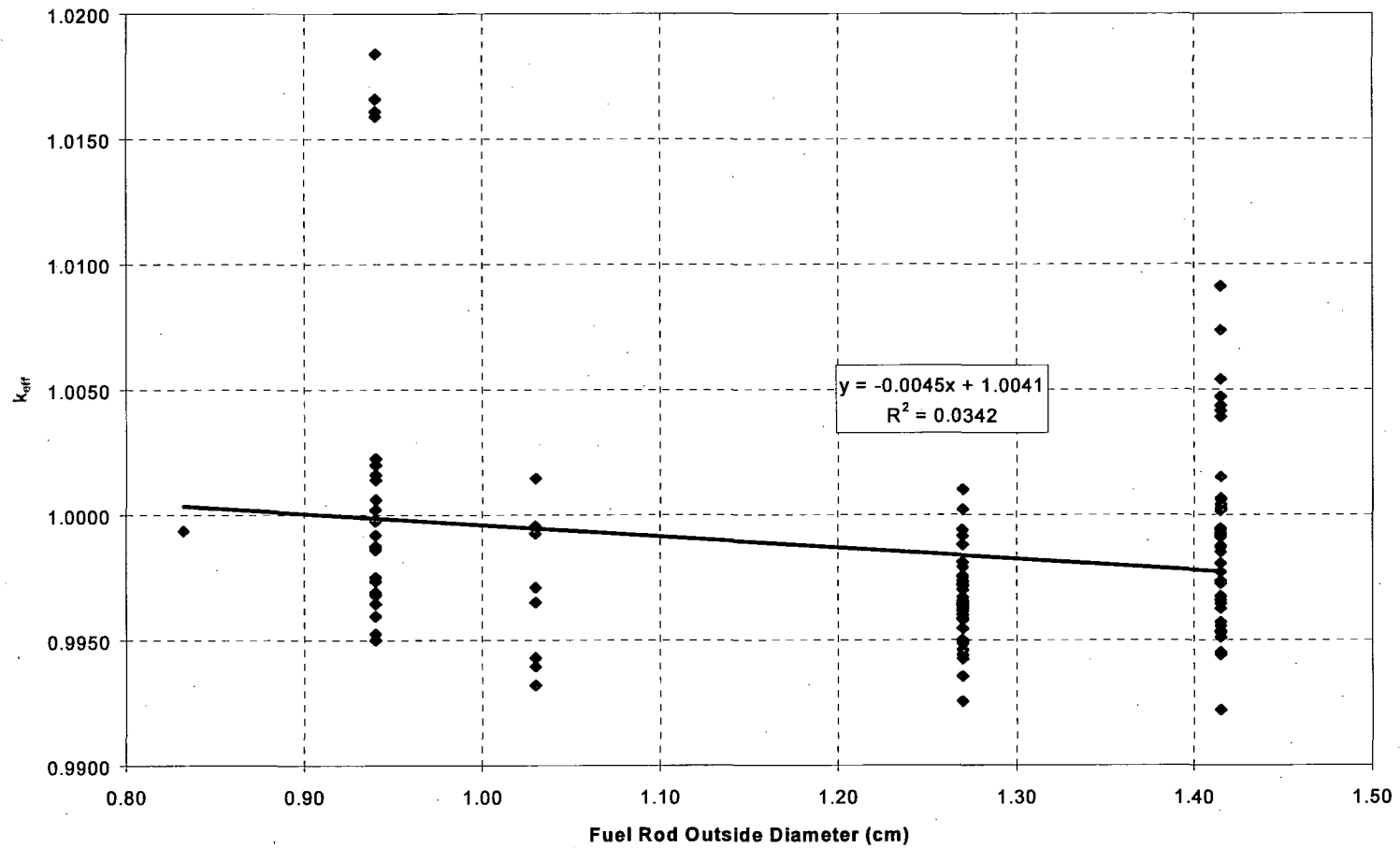


Figure 6.5.2-9 USLSTATS Output - k_{eff} versus Gap Thickness

uslstats: a utility to calculate upper subcritical
limits for criticality safety applications

Version 1.3.4, February 12, 1998
Oak Ridge National Laboratory

Input to statistical treatment from file:gap_keff.txt

Title: 110 LWR CRITICAL EXPERIMENT KEFF VS GAP THICKNESS

Proportion of the population = .995
Confidence of fit = .950
Confidence on proportion = .950
Number of observations = 54
Minimum value of closed band = 0.00
Maximum value of closed band = 0.00
Administrative margin = 0.05

independent variable - x	dependent variable - y	deviation in y	independent variable - x	dependent variable - y	deviation in y
6.33000E+00	9.96400E-01	1.00000E-03	1.05100E+01	9.96200E-01	1.00000E-03
9.03000E+00	9.97600E-01	1.00000E-03	1.10900E+01	9.95500E-01	1.00000E-03
1.04400E+01	9.95000E-01	1.00000E-03	1.31900E+01	9.97200E-01	1.00000E-03
1.14700E+01	9.96000E-01	1.00000E-03	1.33700E+01	9.97400E-01	1.00000E-03
7.56000E+00	9.95000E-01	1.00000E-03	1.29600E+01	9.96700E-01	1.00000E-03
9.62000E+00	9.95500E-01	1.00000E-03	9.95000E+00	9.94400E-01	1.00000E-03
7.36000E+00	9.96300E-01	1.00000E-03	7.82000E+00	9.94600E-01	1.00000E-03
9.52000E+00	9.96700E-01	1.00000E-03	9.88800E+00	9.97500E-01	1.00000E-03
1.19200E+01	9.95900E-01	1.00000E-03	1.04380E+01	9.97000E-01	1.00000E-03
1.06200E+01	9.94400E-01	1.00000E-03	1.04380E+01	9.95800E-01	1.00000E-03
8.58000E+00	9.95700E-01	1.00000E-03	9.59800E+00	9.96400E-01	1.00000E-03
9.65000E+00	9.95300E-01	1.00000E-03	8.74800E+00	9.95500E-01	1.00000E-03
6.10000E+00	9.96600E-01	1.00000E-03	8.56600E+00	9.94300E-01	1.00000E-03
8.08000E+00	9.95600E-01	1.00000E-03	9.16600E+00	9.97200E-01	1.00000E-03
5.76000E+00	9.96800E-01	1.00000E-03	9.09600E+00	9.98800E-01	1.00000E-03
7.90000E+00	9.97300E-01	1.00000E-03	9.24600E+00	9.96500E-01	1.00000E-03
6.72000E+00	9.96300E-01	1.00000E-03	8.86600E+00	9.96600E-01	1.00000E-03
0.00000E+00	1.00150E+00	1.00000E-03	8.64600E+00	9.95900E-01	1.00000E-03
1.64000E+00	9.99300E-01	1.00000E-03	8.12600E+00	9.94300E-01	1.00000E-03
1.64000E+00	9.97100E-01	1.00000E-03	7.25600E+00	9.94900E-01	1.00000E-03
1.64000E+00	9.99600E-01	1.00000E-03	9.64600E+00	9.99200E-01	1.00000E-03
3.27000E+00	9.96500E-01	1.00000E-03	9.69600E+00	1.00020E+00	1.00000E-03
3.27000E+00	9.94000E-01	1.00000E-03	8.08600E+00	9.97900E-01	1.00000E-03
4.91000E+00	9.93200E-01	1.00000E-03	7.64600E+00	9.92600E-01	1.00000E-03
6.54000E+00	9.94300E-01	1.00000E-03	9.08600E+00	9.93600E-01	1.00000E-03
1.31000E+01	1.00100E+00	1.00000E-03	9.41600E+00	9.98100E-01	1.00000E-03
1.29800E+01	9.99400E-01	1.00000E-03	9.77600E+00	9.96400E-01	1.00000E-03

chi = 1.7407 (upper bound = 9.49). The data tests normal.

Figure 6.5.2-9 USLSTATS Output - k_{eff} versus Gap Thickness (continued)

Output from statistical treatment

110 LWR CRITICAL EXPERIMENT KEFF VS GAP THICKNESS

Number of data points (n)	54
Linear regression, $k(X)$	$0.9968 + (-3.5885E-05)*X$
Confidence on fit (1-gamma) [input]	95.0%
Confidence on proportion (alpha) [input]	95.0%
Proportion of population falling above lower tolerance interval (rho) [input]	99.5%
Minimum value of X	0.0000
Maximum value of X	13.3700
Average value of X	8.44389
Average value of k	0.99646
Minimum value of k	0.99260
Variance of fit, $s(k,X)^2$	3.6340E-06
Within variance, $s(w)^2$	1.0000E-06
Pooled variance, $s(p)^2$	4.6340E-06
Pooled std. deviation, $s(p)$	2.1527E-03
$C(\alpha, \rho) * s(p)$	8.6255E-03
student-t @ $(n-2, 1-\gamma)$	1.67620E+00
Confidence band width, W	3.8972E-03
Minimum margin of subcriticality, $C*s(p)-W$	4.7283E-03

Upper subcritical limits: (0.00000 <= X <= 13.37000)

USL Method 1 (Confidence Band with Administrative Margin) $USL1 = 0.9429 + (-3.5885E-05)*X$

USL Method 2 (Single-Sided Uniform Width Closed Interval Approach) $USL2 = 0.9881 + (-3.5885E-05)*X$

USLs Evaluated Over Range of Parameter X:

X: 0.00 1.91 3.82 5.73 7.64 9.55 11.46 13.37

USL-1: 0.9429 0.9428 0.9427 0.9427 0.9426 0.9425 0.9425 0.9424

USL-2: 0.9881 0.9881 0.9880 0.9879 0.9879 0.9878 0.9877 0.9877

Thus spake USLSTATS

Finis.

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
1.01	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Boral	0.0720	6.33	Inf	Water	1.00E-07	0.9964	0.0010
1.02	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Boral	0.0720	9.03	Inf	Water	9.95E-08	0.9976	0.0010
1.03	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (no boron)	0	10.44	Inf	Water	9.97E-08	0.9950	0.0010
1.04	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (no boron)	0	11.47	Inf	Water	9.95E-08	0.9960	0.0010
1.05	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (1.05% boron)	0.0049	7.56	Inf	Water	1.02E-07	0.9950	0.0010
1.06	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (1.05% boron)	0.0049	9.62	Inf	Water	1.01E-07	0.9955	0.0010
1.07	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (1.62% boron)	0.0074	7.36	Inf	Water	1.02E-07	0.9963	0.0010
1.08	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	304L Steel (1.62% boron)	0.0074	9.52	Inf	Water	9.99E-08	0.9967	0.0010
1.09	3 clusters; 20x17 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	None	Na	11.92	Inf	Water	1.01E-07	0.9959	0.0010
2.01	1.26 (square)	4.75	1.26	0.79	0.94	Al	98.21	0	Na	Na	Na	Na	Water	2.57E-07	0.9960	0.0010
2.02	1.60 (square)	4.75	1.60	0.79	0.94	Al	217.26	0	Na	Na	Na	Na	Water	1.15E-07	0.9968	0.0010
2.03	2.10 (square)	4.75	2.10	0.79	0.94	Al	443.75	0	Na	Na	Na	Na	Water	7.31E-08	0.9960	0.0010
2.04	1.35 (triangular)	4.75	1.35	0.79	0.94	Al	97.08	0	Na	Na	Na	Na	Water	2.80E-07	0.9953	0.0010
2.05	1.72 (triangular)	4.75	1.72	0.79	0.94	Al	217.51	0	Na	Na	Na	Na	Water	1.15E-07	0.9975	0.0010
2.06	2.26 (triangular)	4.75	2.26	0.79	0.94	Al	445.38	0	Na	Na	Na	Na	Water	7.34E-08	0.9965	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
2.07	1.26 (square-1 in 5 missing)	4.75	1.26	0.79	0.94	Al	97.08	0	Na	Na	Na	Na	Water	2.65E-07	0.9950	0.0010
2.08	1.26 (square-1 in 2 missing)	4.75	1.26	0.79	0.94	Al	217.51	0	Na	Na	Na	Na	Water	1.16E-07	0.9974	0.0010
2.09	1.26 (square-1 in 3 missing)	4.75	1.26	0.79	0.94	Al	445.38	0	Na	Na	Na	Na	Water	7.42E-08	0.9969	0.0010
3.01	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	None	Na	10.62	Inf	Water	1.18E-07	0.9944	0.0010
3.02	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (no boron)	0	8.58	Inf	Water	1.17E-07	0.9957	0.0010
3.03	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (no boron)	0	9.65	Inf	Water	1.18E-07	0.9953	0.0010
3.04	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (1.05% boron)	0.0049	6.10	Inf	Water	1.19E-07	0.9966	0.0010
3.05	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (1.05% boron)	0.0049	8.08	Inf	Water	1.18E-07	0.9956	0.0010
3.06	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (1.62% boron)	0.0074	5.76	Inf	Water	1.18E-07	0.9968	0.0010
3.07	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	304L Steel (1.62% boron)	0.0074	7.90	Inf	Water	1.16E-07	0.9973	0.0010
3.08	3 clusters; 8x15 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Boral	0.0720	6.72	Inf	Water	1.19E-07	0.9963	0.0010
7.01	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	1037	None	Na	0	Inf	Water	2.56E-07	1.0015	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
7.02	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	769	None	Na	1.64	Inf	Water	2.05E-07	0.9993	0.0010
7.03	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	0	B ₄ C Pins	Na	1.64	Inf	Water	2.03E-07	0.9971	0.0010
7.04	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	15	B/Al (1.61wt% B)	0.0052	1.64	Inf	Water	1.98E-07	0.9996	0.0010
7.05	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	217	Stainless Steel	0	3.27	Inf	Water	1.75E-07	0.9965	0.0010
7.06	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	320	B/Al (0.1wt% B)	0.0003	3.27	Inf	Water	1.79E-07	0.9940	0.0010
7.07	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	72	B/Al (0.1wt% B)	0.0003	4.91	Inf	Water	1.61E-07	0.9932	0.0010
7.08	3x3 clusters; 14x14 rods	2.46	1.6358	1.206	1.03	Al	453.84	0	None	Na	6.54	Inf	Water	1.44E-07	0.9943	0.0010
27.01	Cylindrical	7.00	1.32	0.743	0.8324	SS	118.39	0	Na	Na	Na	Na	Water	2.09E-07	0.9994	0.0010
32.01	14x14 array	4.74	1.60	0.79	0.94	Al	217.31	0	Na	Na	Na	0.0	Lead and light water	1.32E-07	1.0161	0.0010
32.02	14x14 array	4.74	1.60	0.79	0.94	Al	217.31	0	Na	Na	Na	0.5	Lead and light water	1.26E-07	1.0184	0.0010
32.03	14x14 array	4.74	1.60	0.79	0.94	Al	217.31	0	Na	Na	Na	1.0	Lead and light water	1.18E-07	1.0166	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
32.04	14x14 array	4.74	1.60	0.79	0.94	Al		0	Na	Na	Na	1.5	Lead and light water	1.14E-07	1.0159	0.0010
40.01	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.0978	Na	Water	1.33E-07	0.9992	0.0010
40.02	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.1956	Na	Water	1.34E-07	1.0002	0.0010
40.03	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.2934	Na	Water	1.33E-07	0.9998	0.0010
40.04	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.3912	Na	Water	1.34E-07	0.9986	0.0010
40.05	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.489	Na	Water	1.37E-07	1.0020	0.0010
40.06	21x21	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.0978	Na	Water	1.36E-07	1.0014	0.0010
40.07	20x21	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.0978	Na	Water	1.34E-07	1.0023	0.0010
40.08	20x20	4.74	1.60	0.79	0.94	Al	217.31	0	Hafnium plate	Na	0.0978	Na	Water	1.30E-07	0.9987	0.0010
40.09	22x22	4.74	1.60	0.79	0.94	Al	217.31	0	None	Na	-	Na	Water	1.29E-07	0.9998	0.0010
40.10	21x21	4.74	1.60	0.79	0.94	Al	217.31	0	None	Na	-	Na	Water	1.27E-07	1.0006	0.0010
40.11	21x20	4.74	1.60	0.79	0.94	Al	217.31	0	None	Na	-	Na	Water	1.25E-07	1.0016	0.0010
40.12	20x20	4.74	1.60	0.79	0.94	Al	217.31	0	None	Na	-	Na	Water	1.25E-07	0.9988	0.0010
17.01	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	13.100	0.000	Lead	1.03E-07	1.0010	0.0010
17.02	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	12.980	0.660	Lead	1.00E-07	0.9994	0.0010
17.03	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	10.510	2.616	Lead	1.00E-07	0.9962	0.0010
17.04	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	11.090	0.000	Uranium	1.05E-07	0.9955	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
17.05	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	13.190	1.321	Uranium	1.02E-07	0.9972	0.0010
17.06	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	13.370	1.956	Uranium	1.02E-07	0.9974	0.0010
17.07	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	12.960	2.616	Uranium	1.00E-07	0.9967	0.0010
17.08	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	9.950	5.405	Uranium	1.01E-07	0.9944	0.0010
17.09	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	7.820	10.676	Uranium	9.86E-08	0.9946	0.0010
17.10	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	9.888	0.000	Steel	1.03E-07	0.9975	0.0010
17.11	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	10.438	0.660	Steel	1.03E-07	0.9970	0.0010
17.12	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	10.438	1.321	Steel	1.02E-07	0.9958	0.0010
17.13	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	9.598	2.616	Steel	9.91E-08	0.9964	0.0010
17.14	3 clusters; 16x19 rods	2.35	2.032	1.1176	1.27	Al	398.80	0	Na	Na	8.748	3.912	Steel	9.88E-08	0.9955	0.0010
17.15	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	8.566	0.000	Steel	1.89E-07	0.9943	0.0010
17.16	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.166	0.660	Steel	1.83E-07	0.9972	0.0010
17.17	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.096	1.321	Steel	1.75E-07	0.9988	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
17.18	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.246	1.684	Steel	1.77E-07	0.9965	0.0010
17.19	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	8.866	2.344	Steel	1.69E-07	0.9966	0.0010
17.20	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	8.646	3.005	Steel	1.73E-07	0.9959	0.0010
17.21	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	8.126	3.912	Steel	1.66E-07	0.9943	0.0010
17.22	18x25(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	7.256	6.726	Steel	1.65E-07	0.9949	0.0010
17.23	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.646	0.000	Lead	1.76E-07	0.9992	0.0010
17.24	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.696	0.660	Lead	1.74E-07	1.0002	0.0010
17.25	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	8.086	3.276	Lead	1.65E-07	0.9979	0.0010
17.26	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	7.646	0.000	Uranium	1.86E-07	0.9926	0.0010
17.27	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.086	1.321	Uranium	1.78E-07	0.9936	0.0010
17.28	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.416	2.616	Uranium	1.66E-07	0.9981	0.0010
17.29	18x23(center), 18x20(two outer)	2.35	1.684	1.1176	1.27	Al	398.80	0	Na	Na	9.776	3.912	Uranium	1.67E-07	0.9964	0.0010
10.01	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	19.495	0.000	Lead	1.22E-07	1.0091	0.0010
10.02	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	19.655	0.660	Lead	1.20E-07	1.0074	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
10.03	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	17.915	1.321	Lead	1.18E-07	1.0044	0.0010
10.04	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	9.175	5.405	Lead	1.15E-07	0.9945	0.0010
10.05	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	14.255	0.000	Uranium	1.32E-07	0.9951	0.0010
10.06	3 clusters; 8x12 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	14.195	1.956	Uranium	1.18E-07	0.9974	0.0010
10.07	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	16.925	3.912	Uranium	1.18E-07	0.9977	0.0010
10.08	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	12.365	5.405	Uranium	1.16E-07	0.9922	0.0010
10.09	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	11.765	0.000	Steel	1.26E-07	1.0004	0.0010
10.10	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	13.125	0.660	Steel	1.25E-07	0.9981	0.0010
10.11	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	12.995	1.321	Steel	1.20E-07	0.9992	0.0010
10.12	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	11.315	2.616	Steel	1.17E-07	0.9987	0.0010
10.13	3 clusters; 8x13 rods	4.31	2.54	1.265	1.415	Al	256.38	0	Na	Na	8.675	5.405	Steel	1.16E-07	0.9954	0.0010
10.14	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	14.393	0.000	Steel	3.27E-07	1.0002	0.0010
10.15	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	15.263	0.660	Steel	3.16E-07	1.0015	0.0010
10.16	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	15.393	1.321	Steel	3.04E-07	1.0003	0.0010

Table 6.5.2-1 MONK8a – JEF 2.2 Library Validation Statistics (Continued)

Case	Configuration	wt % ²³⁵ U	Pitch (cm)	Fuel OD (cm)	Clad OD (cm)	Clad Mat'l.	H/U (fissile)	Sol. B (ppm)	Poison Type/Absorber	G ¹⁰ B/cm ²	Cluster Gap (cm)	Wall/ Cluster (cm)	Reflector	Mean Log(E) Neutrons Causing Fission	k _{eff} (JEF2.2)	σ
10.17	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	15.363	1.956	Steel	2.97E-07	1.0039	0.0010
10.18	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	14.973	2.616	Steel	2.91E-07	0.9994	0.0010
10.19	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	13.343	5.405	Steel	2.80E-07	1.0007	0.0010
10.20	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	17.263	0.000	Lead	3.11E-07	1.0054	0.0010
10.21	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	17.703	0.660	Lead	3.01E-07	1.0047	0.0010
10.22	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	16.953	1.956	Lead	2.89E-07	1.0042	0.0010
10.23	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	13.873	5.001	Lead	2.81E-07	0.9993	0.0010
10.24	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	14.853	0.000	Uranium	3.33E-07	0.9965	0.0010
10.25	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	16.233	0.660	Uranium	3.23E-07	0.9991	0.0010
10.26	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	17.793	1.321	Uranium	3.14E-07	0.9995	0.0010
10.27	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	18.763	1.956	Uranium	3.05E-07	1.0006	0.0010
10.28	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	18.893	2.616	Uranium	3.01E-07	0.9991	0.0010
10.29	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	18.303	3.276	Uranium	2.88E-07	0.9985	0.0010
10.30	3 clusters; 12x16 rods	4.31	1.892	1.265	1.415	Al	256.38	0	Na	Na	15.923	5.405	Uranium	2.83E-07	0.9988	0.0010

Table 6.5.2-2 Range of Correlated Parameters for Connecticut Yankee Fuel

Parameter	Benchmark Minimum Value	Benchmark Maximum Value	CY-MPC
Enrichment (wt % ^{235}U)	2.35	7.00	4.61
Rod pitch (cm)	1.26	2.54	1.429
H/U (fissile) atomic ratio	97.08	453.84	130.17
^{10}B loading (g/cm^2)	0.000	0.072	0.02
Log energy causing fission	7.31E-08	3.33E-07	2.88E-07
Cluster gap thickness (cm)	0.0	11.92	2.54 to 8.89
Fuel diameter (cm)	0.743	1.265	0.975
Clad diameter (cm)	0.8324	1.4150	1.076

Table 6.5.2-3 MONK8a – Correlation Coefficient for Linear Curve-Fit of Critical Benchmarks

Correlation Studied	Correlation Coefficient (R)
k_{eff} versus enrichment	0.390
k_{eff} versus rod pitch	0.140
k_{eff} versus H/U (fissile) atomic ratio	0.369
k_{eff} versus ^{10}B loading	0.273
k_{eff} versus log energy causing fission	0.127
k_{eff} versus cluster gap thickness	0.532
k_{eff} versus fuel diameter	0.236
k_{eff} versus clad diameter	0.185

6.6 References

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6.7 Supplemental Data

This section presents the input and/or output files for the NAC-STC directly loaded fuel configuration, and for the Yankee-MPC and CY-MPC canistered fuel configurations.

Figures 6.7-1 and 6.7-2 present the CSAS25 input/output files for the criticality analysis of the directly loaded fuel.

Figures 6.7-3 through 6.7-8 present the CSAS25 input/output files for the Yankee Class fuel in the NAC-MPC canistered configuration, including reconfigured Yankee Class fuel and the enlarged fuel tube configuration.

Figure 6.7-9 presents the MONK8a input and output file for the maximum reactivity CY-MPC canistered fuel configuration.

Figure 6.7-10 presents a typical CSAS25 input and output file for the Framatome-Cogema fuel.

Figure 6.7-11 presents a typical CSAS input/output summary for the Damaged Fuel Can configuration for normal conditions.

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis

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PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
NAC-STC 26 DIRECTLY LOADED, WEST 17x17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM
27GROUPNDF4 LATTICECELL
'FUEL
UO2 1 0.95 293.0 92235 4.20 92238 95.80 END
'CLAD
ZIRCALLOY 2 1.0 293.0 END
'H2O CASK INTERIOR
H2O 3 1.0 293.0 END
'AL DISK
AL 4 1.0 293.0 END
'CASK / DISK STEEL
SS304 $ 1.0 293.0 END
'BORAL SHEETS
AL 6 DEN=2.6226 0.5738 293.0 END
B-10 6 DEN=2.6226 0.0450 293.0 END
B-11 6 DEN=2.6226 0.2735 293.0 END
C 6 DEN=2.6226 0.0926 293.0 END
'LEAD SHIELD
PB 7 1.0 293.0 END
'NS4FR SHIELD
B-10 8 0.0 8.553-5 293.0 END
B-11 8 0.0 3.422-4 293.0 END
AL 8 0.0 7.763-3 293.0 END
H 8 0.0 5.854-2 293.0 END
O 8 0.0 2.609-2 293.0 END
C 8 0.0 2.264-2 293.0 END
N 8 0.0 1.394-3 293.0 END
'CASK EXTERIOR WATER
H2O 9 1.0 293.0 END
'PELLET CLAD GAP WATER
H2O 10 1.0 293.0 END
END COMP
SQUAREPITCH 1.2598 0.7844 1 3 0.9144 2 0.8002 0 END
NAC-STC 26 DIRECTLY LOADED, WEST 17x17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM
READ PARAM RUN=YES PLT=NO TME=5000 GEN=803 NPG=1000 END PARAM
READ GEOM
UNIT 1
COM='FUEL PIN CELL - FOR WATER ELEVATION'
CYLINDER 1 1 0.3922 2P2.3749
CYLINDER 0 1 0.4001 2P2.3749
CYLINDER 2 1 0.4572 2P2.3749
CUBOID 3 1 4P0.6299 2P2.3749
UNIT 2
COM='GUIDE/INSTRUMENT TUBE CELL - FOR WATER ELEVATION'
CYLINDER 3 1 0.5715 2P2.3749
CYLINDER 2 1 0.6121 2P2.3749
CUBOID 3 1 4P0.6299 2P2.3749
UNIT 3
COM='FUEL PIN CELL - FOR STEEL DISK ELEVATION'
CYLINDER 1 1 0.3922 2P0.6350
CYLINDER 0 1 0.4001 2P0.6350
CYLINDER 2 1 0.4572 2P0.6350
CUBOID 3 1 4P0.6299 2P0.6350
UNIT 4
COM='GUIDE/INSTRUMENT TUBE CELL - FOR STEEL DISK ELEVATION'
CYLINDER 3 1 0.5715 2P0.6350
CYLINDER 2 1 0.6121 2P0.6350
CUBOID 3 1 4P0.6299 2P0.6350
UNIT 5
COM='FUEL PIN CELL - FOR AL DISK ELEVATION'
CYLINDER 1 1 0.3922 2P0.7938
CYLINDER 0 1 0.4001 2P0.7938
CYLINDER 2 1 0.4572 2P0.7938
CUBOID 3 1 4P0.6299 2P0.7938
UNIT 6
COM='GUIDE/INSTRUMENT TUBE CELL - FOR AL DISK ELEVATION'
CYLINDER 3 1 0.5715 2P0.7938
CYLINDER 2 1 0.6121 2P0.7938
```


Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
CUBOID 3 1 4P0.6299 2P0.7938
UNIT 21
COM='ASSEMBLY - FOR WATER ELEVATION'
ARRAY 1 -10.7083 -10.7083 -2.3749
UNIT 22
COM='ASSEMBLY - FOR STEEL DISK ELEVATION'
ARRAY 2 -10.7083 -10.7083 -0.6350
UNIT 23
COM='ASSEMBLY - FOR AL DISK ELEVATION'
ARRAY 3 -10.7083 -10.7083 -0.7938
UNIT 31
COM='X-X BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P2.3749
CUBOID 4 1 2P10.3886 2P0.0951 2P2.3749
UNIT 32
COM='Y-Y BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P2.3749
CUBOID 4 1 2P0.0951 2P10.3886 2P2.3749
UNIT 33
COM='X-X BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.6350
CUBOID 4 1 2P10.3886 2P0.0951 2P0.6350
UNIT 34
COM='Y-Y BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.6350
CUBOID 4 1 2P0.0951 2P10.3886 2P0.6350
UNIT 35
COM='X-X BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.7938
CUBOID 4 1 2P10.3886 2P0.0951 2P0.7938
UNIT 36
COM='Y-Y BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.7938
CUBOID 4 1 2P0.0951 2P10.3886 2P0.7938
UNIT 40
COM='FUEL TUBE - FOR WATER ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 41
COM='FUEL TUBE - FOR WATER ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 42
COM='FUEL TUBE - FOR WATER ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 43
COM='FUEL TUBE - FOR WATER ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 44
COM='FUEL TUBE - FOR WATER ELEVATION (TL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 45
COM='FUEL TUBE - FOR WATER ELEVATION (TR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 +0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 50
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 51
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 52
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 53
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 54
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TL)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.4803 +0.4803 0.0
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 55
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TR) '
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 +0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 60
COM='FUEL TUBE - FOR AL DISK ELEVATION (B) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 61
COM='FUEL TUBE - FOR AL DISK ELEVATION (T) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 62
COM='FUEL TUBE - FOR AL DISK ELEVATION (BL) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 63
COM='FUEL TUBE - FOR AL DISK ELEVATION (BR) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 64
COM='FUEL TUBE - FOR AL DISK ELEVATION (TL) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 -0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 65
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
COM='FUEL TUBE - FOR AL DISK ELEVATION (TR)'  
CUBOID 3 1 4P11.1887 2P0.7938  
HOLE 23 +0.4803 +0.4803 0.0  
CUBOID 5 1 4P11.3101 2P0.7938  
CUBOID 3 1 4P11.5006 2P0.7938  
HOLE 35 0.0 +11.4054 0.0  
HOLE 35 0.0 -11.4054 0.0  
HOLE 36 +11.4054 0.0 0.0  
HOLE 36 -11.4054 0.0 0.0  
CUBOID 5 1 4P11.5461 2P0.7938  
UNIT 70  
COM='DISK OPENING - FOR WATER ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 40 0.0 -0.1810 0.0  
UNIT 71  
COM='DISK OPENING - FOR WATER ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 41 0.0 +0.1810 0.0  
UNIT 72  
COM='DISK OPENING - FOR WATER ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 42 -0.1810 -0.1810 0.0  
UNIT 73  
COM='DISK OPENING - FOR WATER ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 43 +0.1810 -0.1810 0.0  
UNIT 74  
COM='DISK OPENING - FOR WATER ELEVATION (TL)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 44 -0.1810 +0.1810 0.0  
UNIT 75  
COM='DISK OPENING - FOR WATER ELEVATION (TR)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 45 +0.1810 +0.1810 0.0  
UNIT 80  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 50 0.0 -0.1810 0.0  
UNIT 81  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 51 0.0 +0.1810 0.0  
UNIT 82  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 52 -0.1810 -0.1810 0.0  
UNIT 83  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 53 +0.1810 -0.1810 0.0  
UNIT 84  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TL)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 54 -0.1810 +0.1810 0.0  
UNIT 85  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TR)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 55 +0.1810 +0.1810 0.0  
UNIT 90  
COM='DISK OPENING - FOR AL DISK ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 60 0.0 -0.1810 0.0  
UNIT 91  
COM='DISK OPENING - FOR AL DISK ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 61 0.0 +0.1810 0.0  
UNIT 92  
COM='DISK OPENING - FOR AL DISK ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 62 -0.1810 -0.1810 0.0  
UNIT 93  
COM='DISK OPENING - FOR AL DISK ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 63 +0.1810 -0.1810 0.0
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
UNIT 94
COM='DISK OPENING - FOR AL DISK ELEVATION (TL)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 64 -0.1810 +0.1810 0.0
UNIT 95
COM='DISK OPENING - FOR AL DISK ELEVATION (TR)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 65 +0.1810 +0.1810 0.0
UNIT 101
COM='BASKET STRUCTURE IN TRANSPORT CASK - WATER ELEVATION'
CYLINDER 3 1 +90.17 2P2.3749
HOLE 75 -58.9864 -40.7822 0.0
HOLE 75 -58.9864 -13.5941 0.0
HOLE 73 -58.9864 +13.5941 0.0
HOLE 73 -58.9864 +40.7822 0.0
HOLE 75 -27.1882 -67.9704 0.0
HOLE 75 -27.1882 -40.7822 0.0
HOLE 75 -27.1882 -13.5941 0.0
HOLE 73 -27.1882 +13.5941 0.0
HOLE 73 -27.1882 +40.7822 0.0
HOLE 73 -27.1882 +67.9704 0.0
HOLE 71 0.0 -67.9704 0.0
HOLE 71 0.0 -40.7822 0.0
HOLE 71 0.0 -13.5941 0.0
HOLE 70 0.0 +13.5941 0.0
HOLE 70 0.0 +40.7822 0.0
HOLE 70 0.0 +67.9704 0.0
HOLE 74 +27.1882 -67.9704 0.0
HOLE 74 +27.1882 -40.7822 0.0
HOLE 74 +27.1882 -13.5941 0.0
HOLE 72 +27.1882 +13.5941 0.0
HOLE 72 +27.1882 +40.7822 0.0
HOLE 72 +27.1882 +67.9704 0.0
HOLE 74 +58.9864 -40.7822 0.0
HOLE 74 +58.9864 -13.5941 0.0
HOLE 72 +58.9864 +13.5941 0.0
HOLE 72 +58.9864 +40.7822 0.0
CYLINDER 5 1 +93.98 2P2.3749
CYLINDER 7 1 +103.43 2P2.3749
CYLINDER 5 1 +110.11 2P2.3749
CYLINDER 8 1 +124.12 2P2.3749
CYLINDER 0 1 +124.44 2P2.3749
CYLINDER 5 1 +125.07 2P2.3749
CUBOID 9 1 4P125.07 2P2.3749
UNIT 102
COM='BASKET STRUCTURE IN TRANSPORT CASK - ST DISK ELEVATION'
CYLINDER 5 1 +89.99 2P0.6350
HOLE 85 -58.9864 -40.7822 0.0
HOLE 85 -58.9864 -13.5941 0.0
HOLE 83 -58.9864 +13.5941 0.0
HOLE 83 -58.9864 +40.7822 0.0
HOLE 85 -27.1882 -67.9704 0.0
HOLE 85 -27.1882 -40.7822 0.0
HOLE 85 -27.1882 -13.5941 0.0
HOLE 83 -27.1882 +13.5941 0.0
HOLE 83 -27.1882 +40.7822 0.0
HOLE 83 -27.1882 +67.9704 0.0
HOLE 81 0.0 -67.9704 0.0
HOLE 81 0.0 -40.7822 0.0
HOLE 81 0.0 -13.5941 0.0
HOLE 80 0.0 +13.5941 0.0
HOLE 80 0.0 +40.7822 0.0
HOLE 80 0.0 +67.9704 0.0
HOLE 84 +27.1882 -67.9704 0.0
HOLE 84 +27.1882 -40.7822 0.0
HOLE 84 +27.1882 -13.5941 0.0
HOLE 82 +27.1882 +13.5941 0.0
HOLE 82 +27.1882 +40.7822 0.0
HOLE 82 +27.1882 +67.9704 0.0
HOLE 84 +58.9864 -40.7822 0.0
HOLE 84 +58.9864 -13.5941 0.0
HOLE 82 +58.9864 +13.5941 0.0
HOLE 82 +58.9864 +40.7822 0.0
CYLINDER 3 1 +90.17 2P0.6350
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```

CYLINDER 5 1 +93.98 2P0.6350
CYLINDER 7 1 +103.43 2P0.6350
CYLINDER 5 1 +110.11 2P0.6350
CYLINDER 8 1 +124.12 2P0.6350
CYLINDER 0 1 +124.44 2P0.6350
CYLINDER 5 1 +125.07 2P0.6350
CUBOID 9 1 4P125.07 2P0.6350
UNIT 103
COM='BASKET STRUCTURE IN TRANSPORT CASK - AL DISK ELEVATION'
CYLINDER 4 1 +89.73 2P0.7938
HOLE 95 -58.9864 -40.7822 0.0
HOLE 95 -58.9864 -13.5941 0.0
HOLE 93 -58.9864 +13.5941 0.0
HOLE 93 -58.9864 +40.7822 0.0
HOLE 95 -27.1882 -67.9704 0.0
HOLE 95 -27.1882 -40.7822 0.0
HOLE 95 -27.1882 -13.5941 0.0
HOLE 93 -27.1882 +13.5941 0.0
HOLE 93 -27.1882 +40.7822 0.0
HOLE 93 -27.1882 +67.9704 0.0
HOLE 91 0.0 -67.9704 0.0
HOLE 91 0.0 -40.7822 0.0
HOLE 91 0.0 -13.5941 0.0
HOLE 90 0.0 +13.5941 0.0
HOLE 90 0.0 +40.7822 0.0
HOLE 90 0.0 +67.9704 0.0
HOLE 94 +27.1882 -67.9704 0.0
HOLE 94 +27.1882 -40.7822 0.0
HOLE 94 +27.1882 -13.5941 0.0
HOLE 92 +27.1882 +13.5941 0.0
HOLE 92 +27.1882 +40.7822 0.0
HOLE 92 +27.1882 +67.9704 0.0
HOLE 94 +58.9864 -40.7822 0.0
HOLE 94 +58.9864 -13.5941 0.0
HOLE 92 +58.9864 +13.5941 0.0
HOLE 92 +58.9864 +40.7822 0.0
CYLINDER 3 1 +90.17 2P0.7938
CYLINDER 5 1 +93.98 2P0.7938
CYLINDER 7 1 +103.43 2P0.7938
CYLINDER 5 1 +110.11 2P0.7938
CYLINDER 8 1 +124.12 2P0.7938
CYLINDER 0 1 +124.44 2P0.7938
CYLINDER 5 1 +125.07 2P0.7938
CUBOID 9 1 4P125.07 2P0.7938
GLOBAL UNIT 104
COM='DISK SLICE STACK'
ARRAY 4 -125.07 -125.07 0.0
CUBOID 9 1 4P125.08 12.3573 0.0
END GEOM
READ ARRAY
ARA=1 NUX=17 NUY=17 NUZ=1 FILL
      34R1
      5R1 2 2R1 2 2R1 2 5R1
      3R1 2 9R1 2 3R1
      17R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
      34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
      34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
      17R1
      3R1 2 9R1 2 3R1
      5R1 2 2R1 2 2R1 2 5R1
      34R1
END FILL
ARA=2 NUX=17 NUY=17 NUZ=1 FILL
      34R3
      5R3 4 2R3 4 2R3 4 5R3
      3R3 4 9R3 4 3R3
      17R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3
      34R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3
      34R3

```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
2R3  4  2R3  4  2R3  4  2R3  4  2R3  4  2R3
      17R3
      3R3  4  9R3  4  3R3
      5R3  4  2R3  4  2R3  4  5R3
      34R3
END FILL
ARA=3 NUX=17 NUY=17 NUZ=1 FILL
      34R5
      5R5  6  2R5  6  2R5  6  5R5
      3R5  6  9R5  6  3R5
      17R5
2R5  6  2R5  6  2R5  6  2R5  6  2R5  6  2R5
      34R5
2R5  6  2R5  6  2R5  6  2R5  6  2R5  6  2R5
      34R5
2R5  6  2R5  6  2R5  6  2R5  6  2R5  6  2R5
      17R5
      3R5  6  9R5  6  3R5
      5R5  6  2R5  6  2R5  6  5R5
      34R5
END FILL
ARA=4 NUX=1 NUY=1 NUZ=4 FILL 101 102 101 103 END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIRROR END BOUNDS
READ PLOT
TTL='XY SLICE OF CASK - ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-120.0 YUL=120.0 ZUL=5.5 XLR=120.0 YLR=-120.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='XY SLICE CASK CENTER AREA ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-27.0 YUL=27.0 ZUL=5.5 XLR=27.0 YLR=-27.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
END PLOT
END DATA
```

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
*****  
***** PROGRAM: CSAS *****  
*****  
***** CREATION DATE: 03-08-96 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
*****  
***** PRODUCTION CODE: CSAS *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 01/27/98 *****  
*****  
***** TIME OF EXECUTION: 03:55:50 *****  
*****  
*****  
*****
```

**** PROBLEM PARAMETERS ****

```
LIB 27GROUPNDF4 LIBRARY
MXX      10 MIXTURES
MSC      19 COMPOSITION SPECIFICATIONS
IZM      4 MATERIAL ZONES
GE LATTICECELL GEOMETRY
MORE     0 0/1 DO NOT READ/READ OPTIONAL PARAMETER DATA
MSLN     0 FUEL SOLUTIONS
```

**** PROBLEM COMPOSITION DESCRIPTION ****

SC	UO2	STANDARD COMPOSITION	
MX	1	MIXTURE NO.	
VF	0.9500	VOLUME FRACTION	
ROTH	10.9600	THEORETICAL DENSITY	
NEL	2	NO. ELEMENTS	
ICP	1	0/1 MIXTURE/COMPOUND	
TEMP	293.0	DEG KELVIN	
	92000	1.00	ATOM/MOLECULE
		92235	4.200 WT%
		92238	95.800 WT%
	8016	2.00	ATOMS/MOLECULE

```

' CLAD
      END

```

SC ZIRCALLOY STANDARD COMPOSITION
MX 2 MIXTURE NO.

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
VF      1.0000 VOLUME FRACTION
ROTH    6.5600 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        40302      1.00 ATOM/MOLECULE

'H2O CASK INTERIOR
END

SC H2O      STANDARD COMPOSITION
MX          3 MIXTURE NO.
VF      1.0000 VOLUME FRACTION
ROTH    0.9982 THEORETICAL DENSITY
NEL      2 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        1001      2.00 ATOMS/MOLECULE
        8016      1.00 ATOM/MOLECULE

'AL DISK
END

SC AL      STANDARD COMPOSITION
MX          4 MIXTURE NO.
VF      1.0000 VOLUME FRACTION
ROTH    2.7020 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        13027     1.00 ATOM/MOLECULE

'CASK / DISK STEEL
END

SC SS304    STANDARD COMPOSITION
MX          5 MIXTURE NO.
VF      1.0000 VOLUME FRACTION
ROTH    7.9200 THEORETICAL DENSITY
NEL      4 NO. ELEMENTS
ICP      0 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        24304     19.000 WT%
        25055      2.000 WT%
        26304     69.500 WT%
        28304      9.500 WT%

'BORAL SHEETS
END

SC AL      STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF      0.5738 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        13027     1.00 ATOM/MOLECULE
END

SC B-10     STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF      0.0450 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5010      1.00 ATOM/MOLECULE
END

SC B-11     STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF      0.2735 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           5011      1.00 ATOM/MOLECULE
END

SC C          STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF          0.0926 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           6012      1.00 ATOM/MOLECULE

'LEAD SHIELD
END

SC PB          STANDARD COMPOSITION
MX          7 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        11.3440 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           82000     1.00 ATOM/MOLECULE

'NS4FR SHIELD
END

SC B-10        STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN         8.5530E-05 ATOMIC DENSITY
ROTH        1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           5010      1.00 ATOM/MOLECULE
END

SC B-11        STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN         3.4220E-04 ATOMIC DENSITY
ROTH        1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           5011      1.00 ATOM/MOLECULE
END

SC AL          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN         7.7630E-03 ATOMIC DENSITY
ROTH        2.7020 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           13027     1.00 ATOM/MOLECULE
END

SC H          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN         5.8540E-02 ATOMIC DENSITY
ROTH        1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           1001      1.00 ATOM/MOLECULE
END

SC O          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN         2.6090E-02 ATOMIC DENSITY
ROTH        1.0000 THEORETICAL DENSITY
```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP     293.0 DEG KELVIN
          8016      1.00 ATOM/MOLECULE
END

SC C      STANDARD COMPOSITION
MX        8 MIXTURE NO.
DEN       2.2640E-02 ATOMIC DENSITY
ROTH      2.1000 THEORETICAL DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
          6012      1.00 ATOM/MOLECULE
END

SC N      STANDARD COMPOSITION
MX        8 MIXTURE NO.
DEN       1.3940E-03 ATOMIC DENSITY
ROTH      1.0000 THEORETICAL DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
          7014      1.00 ATOM/MOLECULE

'CASK EXTERIOR WATER
END

SC H2O     STANDARD COMPOSITION
MX         9 MIXTURE NO.
VF         1.0000 VOLUME FRACTION
ROTH       0.9982 THEORETICAL DENSITY
NEL         2 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP       293.0 DEG KELVIN
          1001      2.00 ATOMS/MOLECULE
          8016      1.00 ATOM/MOLECULE

'PELLET CLAD GAP WATER
END

SC H2O     STANDARD COMPOSITION
MX        10 MIXTURE NO.
VF         1.0000 VOLUME FRACTION
ROTH       0.9982 THEORETICAL DENSITY
NEL         2 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP       293.0 DEG KELVIN
          1001      2.00 ATOMS/MOLECULE
          8016      1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH     1.2598 CM CENTER TO CENTER SPACING
FUELOD    0.7844 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL      1 MIXTURE NO. OF FUEL
MMOD       3 MIXTURE NO. OF MODERATOR
CLADOD    0.9144 CM CLAD OUTER DIAMETER
MCLAD      2 MIXTURE NO. OF CLAD
GAPOD     0.8002 CM GAP OUTER DIAMETER
MGAP       0 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD
```

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
*****  
*****  
***** PROGRAM: O0O002 *****  
*****  
***** CREATION DATE: 09-28-95 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
*****  
***** PRODUCTION CODE: NITAWL *****  
*****  
***** VERSION: 3.0 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 01/27/98 *****  
*****  
***** TIME OF EXECUTION: 03:55:53 *****  
*****  
*****  
*****
```

08/12/94

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

NUCLIDES FROM XSDRN TAPE					
1	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	
2	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	
3	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	
4	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	10001001	
5	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010	
6	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010	
7	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011	
8	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011	
9	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012	
10	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012	
11	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014	
12	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016	
13	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	3008016	
14	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	8008016	
15	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	9008016	
16	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	10008016	
17	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	4013027	
18	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	6013027	
19	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	8013027	
20	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5024304	
21	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94	5025055	
22	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5026304	
23	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5028304	
24	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94	2040302	
25	PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94	7082000	
26	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94	1092235	
27	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94	1092238	
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	10001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	3008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	8008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	9008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT		TEMPERATURE=	293.00

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 10008016 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 4013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 6013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 8013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5024304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

MANGANESE-55 ENDF/B-IV MAT 1197 UPDATED 08/12/94 5025055 TEMPERATURE= 293.00

GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	54.466	TEMPERATURE(KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	2.590	LUMPED NUCLEAR DENSITY	=	1.7363295E-03
SPIN FACTOR (G)	=	14.448	LUMP DIMENSION (A-BAR)	=	0.0000000E+00
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	0.0000000E+00

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 55.845 SIGMA(PER ABSORBER ATOM)= 3.4663022E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 55.925 SIGMA(PER ABSORBER ATOM)= 1.2557598E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-5.518788E-04	0.000000E+00	-3.944190E-01
9	-2.797993E-03	0.000000E+00	-2.293471E+00
10	-3.291452E-01	0.000000E+00	-3.820862E+01
11	-2.680562E+00	0.000000E+00	-1.159996E+02

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	3.33719E+00
FISSION	0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5026304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5028304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

ZIRCALLOY ENDF/B-IV MAT 1284 UPDATED 08/12/94 2040302 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	90.436	TEMPERATURE(KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	6.385	LUMPED NUCLEAR DENSITY	=	4.3307818E-02
SPIN FACTOR (G)	=	1.079	LUMP DIMENSION (A-BAR)	=	4.5719999E-01

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

INNER RADIUS = 4.0009999E-01 DANCOFF CORRECTION (C) = 4.5673078E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-3.353624E-04	0.000000E+00	-2.667271E-01
9	-2.345404E-02	0.000000E+00	-9.666542E-01
10	-2.364268E-02	0.000000E+00	-5.427781E-01
11	-8.133699E-02	0.000000E+00	-3.886799E-01

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	5.59244E-01
FISSION	0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

PB 1288 218NGP 042375 P-3 293K

UPDATED 08/12/94 7082000 TEMPERATURE= 293.00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-235 ENDF/B-IV MAT 1261

UPDATED 08/12/94 1092235 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	233.025	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	11.500	LUMPED NUCLEAR DENSITY	=	9.8766887E-04
SPIN FACTOR (G)	=	15171.100	LUMP DIMENSION (A-BAR)	=	3.9219999E-01
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	2.3525730E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA (PER ABSORBER ATOM) = 1.8285364E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 238.051 SIGMA (PER ABSORBER ATOM) = 2.7760712E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
12	-3.904891E+00	-2.400683E+00	-9.365656E-02
13	-1.223861E+01	-5.975270E+00	-2.623999E-01
14	-9.154386E+00	-5.417213E+00	-6.158343E-02
15	-5.171851E-04	-3.930799E-04	4.654119E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	1.98617E+02
FISSION	1.19162E+02

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-238 ENDF/B-IV MAT 1262

UPDATED 08/12/94 1092238 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	236.006	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	10.599	LUMPED NUCLEAR DENSITY	=	2.2243705E-02
SPIN FACTOR (G)	=	656.527	LUMP DIMENSION (A-BAR)	=	3.9219999E-01

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

INNER RADIUS = 0.0000000E+00 DANCOFF CORRECTION (C) = 2.3525730E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA(PER ABSORBER ATOM)= 8.1190996E+00

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 235.044 SIGMA(PER ABSORBER ATOM)= 5.2849644E-01

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
9	-4.655258E-02	0.000000E+00	-4.609913E-01
10	-1.139702E+00	-2.338724E-05	-6.895151E+00
11	-9.892670E+00	0.000000E+00	-2.712805E+01
12	-4.299526E+01	0.000000E+00	-4.994605E+01
13	-5.378721E+01	0.000000E+00	-1.765308E+01
14	-1.041276E+02	0.000000E+00	-6.054823E+00
15	-8.495994E-07	0.000000E+00	1.647857E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 1.88041E+01
FISSION 4.93936E-04

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

[illegible]

```

***
***                                     NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM
***
-----
***                                ***** NUMERIC PARAMETERS *****                                ***
***
*** TME MAXIMUM PROBLEM TIME (MIN) *****
***
*** TBA TIME PER GENERATION (MIN) 0.50
***
*** GEN NUMBER OF GENERATIONS 803
***
*** NPG NUMBER PER GENERATION 1000
***
*** NSK NUMBER OF GENERATIONS TO BE SKIPPED 3
***
*** BEG BEGINNING GENERATION NUMBER 1
***
*** RES GENERATIONS BETWEEN CHECKPOINTS 0
***
*** X1D NUMBER OF EXTRA 1-D CROSS SECTIONS 1
***
*** NBK NEUTRON BANK SIZE 1025
***
*** XNB EXTRA POSITIONS IN NEUTRON BANK 0
***
*** NFB FISSION BANK SIZE 1000
***
*** XFB EXTRA POSITIONS IN FISSION BANK 0
***
*** WTA DEFAULT VALUE OF WEIGHT AVERAGE 0.5000
***

```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```

***          WTH          WEIGHT HIGH FOR SPLITTING          3.0000          ***
***
***          WTL          WEIGHT LOW FOR RUSSIAN ROULETTE      0.3333          ***
***
***          RND          STARTING RANDOM NUMBER              BB827100001          ***
***
***          NB8          NUMBER OF D.A. BLOCKS ON UNIT 8      200          ***
***
***          NL8          LENGTH OF D.A. BLOCKS ON UNIT 8      512          ***
***
***          ADJ          MODE OF CALCULATION                  FORWARD          ***
***
***          INPUT DATA WRITTEN ON RESTART UNIT              NO          ***
***
***          BINARY DATA INTERFACE                            YES          ***
***
*****
***
***          NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM          ***
***
*****
***          LOGICAL PARAMETERS          ***
***
*** RUN EXECUTE PROBLEM AFTER CHECKING DATA YES          PLT PLOT PICTURE MAP(S)          NO ***
***
*** FLX COMPUTE FLUX          NO          FDN COMPUTE FISSION DENSITIES          NO ***
***
*** SMU COMPUTE AVG UNIT SELF-MULTIPLICATION          NO          NUB COMPUTE NU-BAR & AVG FISSION GROUP          YES ***
***
*** MKU COMPUTE MATRIX K-EFF BY UNIT NUMBER          NO          MKP COMPUTE MATRIX K-EFF BY UNIT LOCATION          NO ***
***
*** CKU COMPUTE COFACTOR K-EFF BY UNIT NUMBER          NO          CKP COMPUTE COFACTOR K-EFF BY UNIT LOCATION          NO ***
***
*** FMU PRINT FISS PROD MATRIX BY UNIT NUMBER          NO          FMP PRINT FISS PROD MATRIX BY UNIT LOCATION          NO ***
***
*** MKH COMPUTE MATRIX K-EFF BY HOLE NUMBER          NO          MKA COMPUTE MATRIX K-EFF BY ARRAY NUMBER          NO ***
***
*** CKH COMPUTE COFACTOR K-EFF BY HOLE NUMBER          NO          CKA COMPUTE COFACTOR K-EFF BY ARRAY NUMBER          NO ***
***
*** FMH PRINT FISS PROD MATRIX BY HOLE NUMBER          NO          FMA PRINT FISS PROD MATRIX BY ARRAY NUMBER          NO ***
***
*** HHL COLLECT MATRIX BY HIGHEST HOLE LEVEL          NO          HAL COLLECT MATRIX BY HIGHEST ARRAY LEVEL          NO ***
***
*** AMX PRINT ALL MIXED CROSS SECTIONS          NO          FAR PRINT FIS. AND ABS. BY REGION          NO ***
***
*** XS1 PRINT 1-D MIXTURE X-SECTIONS          NO          GAS PRINT FAR BY GROUP          NO ***
***
*** XS2 PRINT 2-D MIXTURE X-SECTIONS          NO          PAX PRINT XSEC-ALBEDO CORRELATION TABLES          NO ***
***
*** XAP PRINT MIXTURE ANGLES & PROBABILITIES          NO          PWT PRINT WEIGHT AVERAGE ARRAY          NO ***
***
*** PKI PRINT FISSION SPECTRUM          NO          PGM PRINT INPUT GEOMETRY          NO ***
***
*** P1D PRINT EXTRA 1-D CROSS SECTIONS          NO          BUG PRINT DEBUG INFORMATION          NO ***
***
***          TRK PRINT TRACKING INFORMATION          NO ***
***
*****
***
***          PARAMETER INPUT COMPLETED

```

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD = 3.0E-05

MIXTURE = 1		DENSITY(G/CC) = 10.412					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
1008016	4.64627E-02	1.18489E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
1092235	9.87669E-04	3.70234E-02	92235	235.0441	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED
08/12/94							
1092238	2.22437E-02	8.44487E-01	92238	238.0510	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED
08/12/94							
MIXTURE = 2		DENSITY(G/CC) = 6.5600					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
2040302	4.33078E-02	1.00000E+00	40000	91.2196	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED
08/12/94							
MIXTURE = 3		DENSITY(G/CC) = 0.99817					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
3001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
3008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
MIXTURE = 4		DENSITY(G/CC) = 2.7020					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
4013027	6.03066E-02	1.00000E+00	13027	26.9818	AL-27 1193 218 GP 040375(5)		UPDATED
08/12/94							
MIXTURE = 5		DENSITY(G/CC) = 7.9200					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
5024304	1.74286E-02	1.90000E-01	24000	51.9957	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
5025055	1.73633E-03	1.99999E-02	25055	54.9379	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED
08/12/94							
5026304	5.93579E-02	6.95000E-01	26000	55.8447	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
5028304	7.72070E-03	9.50001E-02	28000	58.6872	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
MIXTURE = 6		DENSITY(G/CC) = 2.5830					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
6005010	7.09799E-03	4.56901E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K		UPDATED
08/12/94							
6005011	3.92356E-02	2.77698E-01	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160	UPDATED
08/12/94							
6006012	1.21874E-02	9.40196E-02	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94							
6013027	3.35871E-02	5.82592E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)		UPDATED
08/12/94							
MIXTURE = 7		DENSITY(G/CC) = 11.344					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K		UPDATED
08/12/94							
MIXTURE = 8		DENSITY(G/CC) = 1.6298					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
8001001	5.85400E-02	6.01023E-02	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
8005010	8.55300E-05	8.72589E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K		UPDATED
08/12/94							
8005011	3.42200E-04	3.83863E-03	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160	UPDATED
08/12/94							
8006012	2.26400E-02	2.76813E-01	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94							
8007014	1.39400E-03	1.98893E-02	7014	14.0033	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED
08/12/94							

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

8008016	2.60900E-02	4.25068E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
8013027	7.76300E-03	2.13416E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)		UPDATED
08/12/94							
MIXTURE =	9	DENSITY(G/CC) =	0.99817				
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
9001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
9008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
MIXTURE =	10	DENSITY(G/CC) =	0.99817				
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
10001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
10008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
			3001001		HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			8001001		HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			9001001		HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			10001001		HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			6005010		B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
			8005010		B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
			6005011		BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
			8005011		BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
			6006012		CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
			8006012		CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
			8007014		NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94
			1008016		OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
			3008016		OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
			8008016		OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
			9008016		OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
			10008016		OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
			4013027		AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
			6013027		AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
			8013027		AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
			5024304		CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
			5025055		MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94
			5026304		FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
			5028304		NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
			2040302		ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94
			7082000		PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94
			1092235		URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94
			1092238		URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94
KENO MESSAGE NUMBER K5-222	1	TRANSFERS FOR MIXTURE	3	WERE CORRECTED FOR BAD MOMENTS.			
KENO MESSAGE NUMBER K5-222	1	TRANSFERS FOR MIXTURE	9	WERE CORRECTED FOR BAD MOMENTS.			
KENO MESSAGE NUMBER K5-222	1	TRANSFERS FOR MIXTURE	10	WERE CORRECTED FOR BAD MOMENTS.			
	0	IO'S WERE USED MIXING CROSS-SECTIONS					
	1-D CROSS SECTION ARRAY ID NUMBERS						
	1	2002 1452	27	18 1018			
	0	IO'S WERE USED PREPARING THE CROSS SECTIONS					

***	NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM						***
***							***
***							***
***	***** ADDITIONAL INFORMATION *****						***
***							***
***	NUMBER OF ENERGY GROUPS	27	USE LATTICE GEOMETRY	YES			***
***	NO. OF FISSION SPECTRUM SOURCE GROUP	1	GLOBAL ARRAY NUMBER	4			***

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

```

***
*** NO. OF SCATTERING ANGLES IN XSECS      2      NUMBER OF UNITS IN THE GLOBAL X DIR.      1 ***
***
*** ENTRIES/NEUTRON IN THE NEUTRON BANK 25      NUMBER OF UNITS IN THE GLOBAL Y DIR.      1 ***
***
*** ENTRIES/NEUTRON IN THE FISSION BANK 18      NUMBER OF UNITS IN THE GLOBAL Z DIR.      4 ***
***
*** NUMBER OF MIXTURES USED      9      USE A GLOBAL REFLECTOR      YES ***
***
*** NUMBER OF BIAS ID'S USED      1      USE NESTED HOLES      YES ***
***
*** NUMBER OF DIFFERENTIAL ALBEDOS USED      0      NUMBER OF HOLES      186 ***
***
*** TOTAL INPUT GEOMETRY REGIONS      154      MAXIMUM HOLE NESTING LEVEL      3 ***
***
*** NUMBER OF GEOMETRY REGIONS USED      154      USE NESTED ARRAYS      YES ***
***
*** LARGEST GEOMETRY UNIT NUMBER      104      NUMBER OF ARRAYS USED      4 ***
***
*** LARGEST ARRAY NUMBER      4      MAXIMUM ARRAY NESTING LEVEL      2 ***
***
***
*** +X BOUNDARY CONDITION      MIRROR      -X BOUNDARY CONDITION      MIRROR ***
***
*** +Y BOUNDARY CONDITION      MIRROR      -Y BOUNDARY CONDITION      MIRROR ***
***
*** +Z BOUNDARY CONDITION      PER      -Z BOUNDARY CONDITION      PER ***
***

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NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM

GENERATION	K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	974	INDEPENDENT	FISSION POINTS WERE GENERATED		
1	8.44990E-01	1.12667E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	980	INDEPENDENT	FISSION POINTS WERE GENERATED		
2	8.68154E-01	1.69333E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
3	8.96286E-01	2.23333E-01	8.96286E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	8.65434E-01	2.76500E-01	8.80860E-01	1.54261E-02	0.00000E+00	0.00000E+00
5	8.97554E-01	3.30500E-01	8.86425E-01	1.05018E-02	0.00000E+00	0.00000E+00
6	9.14550E-01	3.81667E-01	8.93456E-01	1.02267E-02	0.00000E+00	0.00000E+00
7	9.00364E-01	4.35667E-01	8.94838E-01	8.04113E-03	0.00000E+00	0.00000E+00
8	8.85585E-01	4.88833E-01	8.93296E-01	6.74422E-03	0.00000E+00	0.00000E+00
9	9.20117E-01	5.42000E-01	8.97127E-01	6.86807E-03	0.00000E+00	0.00000E+00
10	8.88359E-01	5.97833E-01	8.96031E-01	6.04807E-03	0.00000E+00	0.00000E+00
11	9.24954E-01	6.49000E-01	8.99245E-01	6.22722E-03	0.00000E+00	0.00000E+00
12	9.12174E-01	7.01167E-01	9.00538E-01	5.71788E-03	0.00000E+00	0.00000E+00
13	9.33608E-01	7.52500E-01	9.03544E-01	5.98230E-03	0.00000E+00	0.00000E+00
14	9.18353E-01	8.03667E-01	9.04778E-01	5.59876E-03	0.00000E+00	0.00000E+00
15	9.05712E-01	8.55000E-01	9.04850E-01	5.15061E-03	0.00000E+00	0.00000E+00
16	9.05532E-01	9.09000E-01	9.04899E-01	4.76879E-03	0.00000E+00	0.00000E+00
17	8.83233E-01	9.59333E-01	9.03454E-01	4.66855E-03	0.00000E+00	0.00000E+00
18	8.73386E-01	1.01333E+00	9.01575E-01	4.75421E-03	0.00000E+00	0.00000E+00
19	9.11353E-01	1.06283E+00	9.02150E-01	4.50269E-03	0.00000E+00	0.00000E+00
20	9.15656E-01	1.11317E+00	9.02901E-01	4.31097E-03	0.00000E+00	0.00000E+00
21	9.23252E-01	1.16433E+00	9.03972E-01	4.21610E-03	0.00000E+00	0.00000E+00
22	9.08404E-01	1.21667E+00	9.04193E-01	4.00588E-03	0.00000E+00	0.00000E+00
23	8.74029E-01	1.26883E+00	9.02757E-01	4.07210E-03	0.00000E+00	0.00000E+00
24	8.99533E-01	1.31817E+00	9.02610E-01	3.88536E-03	0.00000E+00	0.00000E+00
25	8.96263E-01	1.37217E+00	9.02334E-01	3.72283E-03	0.00000E+00	0.00000E+00
26	8.99279E-01	1.42250E+00	9.02207E-01	3.56661E-03	0.00000E+00	0.00000E+00
27	9.10017E-01	1.47467E+00	9.02520E-01	3.43521E-03	0.00000E+00	0.00000E+00
28	8.95336E-01	1.52417E+00	9.02243E-01	3.31199E-03	0.00000E+00	0.00000E+00
29	9.74939E-01	1.57450E+00	9.04936E-01	4.17204E-03	0.00000E+00	0.00000E+00
30	9.31477E-01	1.62483E+00	9.05884E-01	4.13052E-03	0.00000E+00	0.00000E+00
31	9.36373E-01	1.67617E+00	9.06935E-01	4.12189E-03	0.00000E+00	0.00000E+00
32	9.52017E-01	1.72833E+00	9.08438E-01	4.25623E-03	0.00000E+00	0.00000E+00
33	9.06217E-01	1.77867E+00	9.08366E-01	4.11726E-03	0.00000E+00	0.00000E+00
34	9.17878E-01	1.82817E+00	9.08663E-01	3.99759E-03	0.00000E+00	0.00000E+00
35	9.35463E-01	1.87567E+00	9.09475E-01	3.95875E-03	0.00000E+00	0.00000E+00
36	9.12519E-01	1.92600E+00	9.09565E-01	3.84160E-03	0.00000E+00	0.00000E+00
37	8.86443E-01	1.97733E+00	9.08904E-01	3.78827E-03	0.00000E+00	0.00000E+00
38	8.67201E-01	2.02667E+00	9.07746E-01	3.85949E-03	0.00000E+00	0.00000E+00
39	9.14923E-01	2.07800E+00	9.07940E-01	3.75874E-03	0.00000E+00	0.00000E+00

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

40	9.27898E-01	2.12833E+00	9.08465E-01	3.69599E-03	0.00000E+00	0.00000E+00
41	8.98449E-01	2.17867E+00	9.08208E-01	3.60912E-03	0.00000E+00	0.00000E+00
42	9.41256E-01	2.22900E+00	9.09034E-01	3.61346E-03	0.00000E+00	0.00000E+00
43	9.26125E-01	2.27750E+00	9.09451E-01	3.54879E-03	0.00000E+00	0.00000E+00
44	9.48744E-01	2.32700E+00	9.10387E-01	3.58740E-03	0.00000E+00	0.00000E+00
...						
766	8.52062E-01	3.88853E+01	9.12782E-01	8.87278E-04	0.00000E+00	0.00000E+00
767	8.89113E-01	3.89367E+01	9.12751E-01	8.86658E-04	0.00000E+00	0.00000E+00
768	9.32841E-01	3.89862E+01	9.12777E-01	8.85888E-04	0.00000E+00	0.00000E+00
769	9.25239E-01	3.90355E+01	9.12794E-01	8.84881E-04	0.00000E+00	0.00000E+00
770	9.17486E-01	3.90858E+01	9.12800E-01	8.83749E-04	0.00000E+00	0.00000E+00
771	9.12554E-01	3.91380E+01	9.12799E-01	8.82599E-04	0.00000E+00	0.00000E+00
772	8.92045E-01	3.91875E+01	9.12772E-01	8.81864E-04	0.00000E+00	0.00000E+00
773	9.13794E-01	3.92378E+01	9.12774E-01	8.80721E-04	0.00000E+00	0.00000E+00
774	9.21233E-01	3.92882E+01	9.12785E-01	8.79648E-04	0.00000E+00	0.00000E+00
775	8.96880E-01	3.93367E+01	9.12764E-01	8.78750E-04	0.00000E+00	0.00000E+00
776	9.37107E-01	3.93880E+01	9.12796E-01	8.78177E-04	0.00000E+00	0.00000E+00
777	9.23805E-01	3.94383E+01	9.12810E-01	8.77158E-04	0.00000E+00	0.00000E+00
778	9.05338E-01	3.94897E+01	9.12800E-01	8.76080E-04	0.00000E+00	0.00000E+00
779	8.91731E-01	3.95382E+01	9.12773E-01	8.75372E-04	0.00000E+00	0.00000E+00
780	9.06924E-01	3.95893E+01	9.12766E-01	8.74278E-04	0.00000E+00	0.00000E+00
781	9.62034E-01	3.96407E+01	9.12829E-01	8.75443E-04	0.00000E+00	0.00000E+00
782	9.16989E-01	3.96892E+01	9.12834E-01	8.74336E-04	0.00000E+00	0.00000E+00
783	9.11415E-01	3.97395E+01	9.12832E-01	8.73218E-04	0.00000E+00	0.00000E+00
784	9.05594E-01	3.97890E+01	9.12823E-01	8.72149E-04	0.00000E+00	0.00000E+00
785	9.14512E-01	3.98393E+01	9.12825E-01	8.71038E-04	0.00000E+00	0.00000E+00
786	9.03778E-01	3.98897E+01	9.12814E-01	8.70002E-04	0.00000E+00	0.00000E+00
787	9.37944E-01	3.99372E+01	9.12846E-01	8.69483E-04	0.00000E+00	0.00000E+00
788	9.56264E-01	3.99875E+01	9.12901E-01	8.70131E-04	0.00000E+00	0.00000E+00
789	9.16645E-01	4.00380E+01	9.12906E-01	8.69038E-04	0.00000E+00	0.00000E+00
790	8.75757E-01	4.00883E+01	9.12859E-01	8.69214E-04	0.00000E+00	0.00000E+00
791	9.29604E-01	4.01423E+01	9.12880E-01	8.68371E-04	0.00000E+00	0.00000E+00
792	9.27283E-01	4.01917E+01	9.12898E-01	8.67462E-04	0.00000E+00	0.00000E+00
793	8.74477E-01	4.02420E+01	9.12849E-01	8.67726E-04	0.00000E+00	0.00000E+00
794	9.09629E-01	4.02923E+01	9.12845E-01	8.66639E-04	0.00000E+00	0.00000E+00
795	9.17668E-01	4.03428E+01	9.12851E-01	8.65567E-04	0.00000E+00	0.00000E+00
796	9.24591E-01	4.03932E+01	9.12866E-01	8.64602E-04	0.00000E+00	0.00000E+00
797	9.03978E-01	4.04443E+01	9.12855E-01	8.63586E-04	0.00000E+00	0.00000E+00
798	9.13778E-01	4.04947E+01	9.12856E-01	8.62502E-04	0.00000E+00	0.00000E+00
799	8.96251E-01	4.05468E+01	9.12835E-01	8.61671E-04	0.00000E+00	0.00000E+00
800	8.95363E-01	4.05963E+01	9.12813E-01	8.60869E-04	0.00000E+00	0.00000E+00
801	9.32395E-01	4.06467E+01	9.12838E-01	8.60140E-04	0.00000E+00	0.00000E+00
802	9.35120E-01	4.06970E+01	9.12866E-01	8.59515E-04	0.00000E+00	0.00000E+00
803	9.34006E-01	4.07473E+01	9.12892E-01	8.58847E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123 EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.
NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM

LIFETIME = 4.01972E-05 + OR - 7.69258E-08 GENERATION TIME = 3.01354E-05 + OR - 4.55607E-08
NU BAR = 2.43862E+00 + OR - 6.74971E-05 AVERAGE FISSION GROUP = 2.22216E+01 + OR - 3.83639E-03
ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 1.94425E-01 + OR - 6.19504E-04

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.91291	+ OR - 0.00086	0.91205 TO 0.91377	0.91119 TO 0.91463	0.91033 TO 0.91549	800000
4	0.91297	+ OR - 0.00086	0.91211 TO 0.91383	0.91126 TO 0.91469	0.91040 TO 0.91555	799000
5	0.91299	+ OR - 0.00086	0.91213 TO 0.91385	0.91127 TO 0.91471	0.91041 TO 0.91557	798000
6	0.91299	+ OR - 0.00086	0.91213 TO 0.91385	0.91127 TO 0.91471	0.91041 TO 0.91557	797000
7	0.91301	+ OR - 0.00086	0.91214 TO 0.91387	0.91128 TO 0.91473	0.91042 TO 0.91559	796000
8	0.91304	+ OR - 0.00086	0.91218 TO 0.91390	0.91132 TO 0.91476	0.91045 TO 0.91563	795000
9	0.91303	+ OR - 0.00086	0.91217 TO 0.91389	0.91131 TO 0.91476	0.91044 TO 0.91562	794000
10	0.91306	+ OR - 0.00086	0.91220 TO 0.91393	0.91134 TO 0.91479	0.91047 TO 0.91565	793000

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

11	0.91305	+ OR - 0.00086	0.91218 TO 0.91391	0.91132 TO 0.91478	0.91045 TO 0.91564	792000
12	0.91305	+ OR - 0.00087	0.91218 TO 0.91391	0.91132 TO 0.91478	0.91045 TO 0.91565	791000
17	0.91307	+ OR - 0.00087	0.91220 TO 0.91394	0.91133 TO 0.91481	0.91046 TO 0.91568	786000
22	0.91311	+ OR - 0.00087	0.91224 TO 0.91399	0.91137 TO 0.91486	0.91049 TO 0.91574	781000
27	0.91323	+ OR - 0.00088	0.91235 TO 0.91410	0.91147 TO 0.91498	0.91059 TO 0.91586	776000
32	0.91307	+ OR - 0.00088	0.91219 TO 0.91394	0.91131 TO 0.91482	0.91044 TO 0.91570	771000
37	0.91307	+ OR - 0.00088	0.91219 TO 0.91396	0.91131 TO 0.91484	0.91043 TO 0.91572	766000
42	0.91309	+ OR - 0.00088	0.91221 TO 0.91398	0.91133 TO 0.91486	0.91044 TO 0.91575	761000
47	0.91293	+ OR - 0.00088	0.91204 TO 0.91381	0.91116 TO 0.91470	0.91027 TO 0.91558	756000
52	0.91280	+ OR - 0.00089	0.91191 TO 0.91369	0.91103 TO 0.91458	0.91014 TO 0.91546	751000
57	0.91276	+ OR - 0.00089	0.91187 TO 0.91365	0.91098 TO 0.91454	0.91008 TO 0.91543	746000
•						
•						
•						
692	0.91252	+ OR - 0.00225	0.91027 TO 0.91476	0.90803 TO 0.91701	0.90578 TO 0.91925	111000
697	0.91267	+ OR - 0.00232	0.91035 TO 0.91498	0.90803 TO 0.91730	0.90571 TO 0.91962	106000
702	0.91140	+ OR - 0.00235	0.90905 TO 0.91374	0.90670 TO 0.91609	0.90435 TO 0.91844	101000
707	0.91289	+ OR - 0.00234	0.91055 TO 0.91522	0.90822 TO 0.91756	0.90588 TO 0.91989	96000
712	0.91244	+ OR - 0.00242	0.91003 TO 0.91486	0.90761 TO 0.91728	0.90519 TO 0.91970	91000
717	0.91205	+ OR - 0.00253	0.90952 TO 0.91457	0.90700 TO 0.91710	0.90447 TO 0.91962	86000
722	0.91162	+ OR - 0.00257	0.90905 TO 0.91419	0.90648 TO 0.91676	0.90391 TO 0.91933	81000
727	0.91275	+ OR - 0.00257	0.91018 TO 0.91532	0.90761 TO 0.91789	0.90504 TO 0.92046	76000
732	0.91330	+ OR - 0.00263	0.91067 TO 0.91594	0.90804 TO 0.91857	0.90541 TO 0.92120	71000
737	0.91244	+ OR - 0.00261	0.90984 TO 0.91505	0.90723 TO 0.91765	0.90463 TO 0.92026	66000
742	0.91297	+ OR - 0.00276	0.91020 TO 0.91573	0.90744 TO 0.91850	0.90467 TO 0.92126	61000
747	0.91369	+ OR - 0.00287	0.91082 TO 0.91656	0.90795 TO 0.91943	0.90508 TO 0.92230	56000
752	0.91364	+ OR - 0.00314	0.91050 TO 0.91677	0.90737 TO 0.91991	0.90423 TO 0.92304	51000
757	0.91452	+ OR - 0.00317	0.91135 TO 0.91770	0.90817 TO 0.92087	0.90500 TO 0.92404	46000
762	0.91332	+ OR - 0.00336	0.90996 TO 0.91667	0.90661 TO 0.92003	0.90325 TO 0.92338	41000
767	0.91589	+ OR - 0.00319	0.91270 TO 0.91908	0.90950 TO 0.92228	0.90631 TO 0.92547	36000
772	0.91587	+ OR - 0.00358	0.91229 TO 0.91944	0.90872 TO 0.92302	0.90514 TO 0.92659	31000
777	0.91535	+ OR - 0.00411	0.91124 TO 0.91946	0.90713 TO 0.92357	0.90302 TO 0.92768	26000
782	0.91505	+ OR - 0.00438	0.91067 TO 0.91943	0.90629 TO 0.92381	0.90191 TO 0.92819	21000
787	0.91518	+ OR - 0.00552	0.90966 TO 0.92069	0.90414 TO 0.92621	0.89862 TO 0.93173	16000
792	0.91248	+ OR - 0.00574	0.90673 TO 0.91822	0.90099 TO 0.92397	0.89525 TO 0.92971	11000
797	0.91782	+ OR - 0.00766	0.91016 TO 0.92548	0.90250 TO 0.93314	0.89484 TO 0.94079	6000

Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM

```

                                FREQUENCY FOR GENERATIONS    4 TO 803
0.8449 TO 0.8500      **
0.8500 TO 0.8552      ****
0.8552 TO 0.8603      **
0.8603 TO 0.8654      *****
0.8654 TO 0.8705      *****
0.8705 TO 0.8756      *****
0.8756 TO 0.8808      *****
0.8808 TO 0.8859      *****
0.8859 TO 0.8910      *****
0.8910 TO 0.8961      *****
0.8961 TO 0.9013      *****
0.9013 TO 0.9064      *****
0.9064 TO 0.9115      *****
0.9115 TO 0.9166      *****
0.9166 TO 0.9218      *****
0.9218 TO 0.9269      *****
0.9269 TO 0.9320      *****
0.9320 TO 0.9371      *****
0.9371 TO 0.9422      *****
0.9422 TO 0.9474      *****
0.9474 TO 0.9525      *****
0.9525 TO 0.9576      *****
0.9576 TO 0.9627      *****
0.9627 TO 0.9679      *****
0.9679 TO 0.9730      **
0.9730 TO 0.9781      ***
0.9781 TO 0.9832      **
0.9832 TO 0.9884      *
```

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                                FREQUENCY FOR GENERATIONS  204 TO 803
0.8449 TO 0.8500      **
0.8500 TO 0.8552      ****
0.8552 TO 0.8603      **
0.8603 TO 0.8654      ***
0.8654 TO 0.8705      *****
0.8705 TO 0.8756      *****
0.8756 TO 0.8808      *****
0.8808 TO 0.8859      *****
0.8859 TO 0.8910      *****
0.8910 TO 0.8961      *****
0.8961 TO 0.9013      *****
0.9013 TO 0.9064      *****
0.9064 TO 0.9115      *****
0.9115 TO 0.9166      *****
0.9166 TO 0.9218      *****
0.9218 TO 0.9269      *****
0.9269 TO 0.9320      *****
0.9320 TO 0.9371      *****
0.9371 TO 0.9422      *****
0.9422 TO 0.9474      *****
0.9474 TO 0.9525      *****
0.9525 TO 0.9576      *****
0.9576 TO 0.9627      *****
0.9627 TO 0.9679      *****
0.9679 TO 0.9730      **
0.9730 TO 0.9781      *
0.9781 TO 0.9832      **
0.9832 TO 0.9884      *
```


Figure 6.7-1 CSAS25 Input/Output for Directly Loaded Fuel Normal Conditions Criticality Analysis (continued)

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, NORMAL LOAD 1.0 1.0, PITCH 250 CM

```

                                FREQUENCY FOR GENERATIONS 404 TO 803
0.8449 TO 0.8500      **
0.8500 TO 0.8552     ***
0.8552 TO 0.8603      *
0.8603 TO 0.8654     ***
0.8654 TO 0.8705     *****
0.8705 TO 0.8756     *****
0.8756 TO 0.8808     *****
0.8808 TO 0.8859     *****
0.8859 TO 0.8910     *****
0.8910 TO 0.8961     *****
0.8961 TO 0.9013     *****
0.9013 TO 0.9064     *****
0.9064 TO 0.9115     *****
0.9115 TO 0.9166     *****
0.9166 TO 0.9218     *****
0.9218 TO 0.9269     *****
0.9269 TO 0.9320     *****
0.9320 TO 0.9371     *****
0.9371 TO 0.9422     *****
0.9422 TO 0.9474     *****
0.9474 TO 0.9525     *****
0.9525 TO 0.9576     *****
0.9576 TO 0.9627     ***
0.9627 TO 0.9679     ***
0.9679 TO 0.9730
0.9730 TO 0.9781      *
0.9781 TO 0.9832     **
0.9832 TO 0.9884      *
```

```

                                FREQUENCY FOR GENERATIONS 604 TO 803
0.8449 TO 0.8500      *
0.8500 TO 0.8552     **
0.8552 TO 0.8603      *
0.8603 TO 0.8654     **
0.8654 TO 0.8705     ***
0.8705 TO 0.8756     **
0.8756 TO 0.8808     *****
0.8808 TO 0.8859     *****
0.8859 TO 0.8910     *****
0.8910 TO 0.8961     *****
0.8961 TO 0.9013     *****
0.9013 TO 0.9064     *****
0.9064 TO 0.9115     *****
0.9115 TO 0.9166     *****
0.9166 TO 0.9218     *****
0.9218 TO 0.9269     *****
0.9269 TO 0.9320     *****
0.9320 TO 0.9371     *****
0.9371 TO 0.9422     *****
0.9422 TO 0.9474     ***
0.9474 TO 0.9525     *****
0.9525 TO 0.9576     *****
0.9576 TO 0.9627     ***
0.9627 TO 0.9679
0.9679 TO 0.9730
0.9730 TO 0.9781      *
0.9781 TO 0.9832
0.9832 TO 0.9884
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis

```
PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
NAC-STC 26 DIRECTLY LOADED, WEST 17x17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25
27GROUPNDF4 LATTICECELL
'FUEL
UO2 1 0.95 293.0 92235 4.20 92238 95.80 END
'CLAD
ZIRCALLOY 2 1.0 293.0 END
'H2O CASK INTERIOR
H2O 3 1.0 293.0 END
'AL DISK
AL 4 1.0 293.0 END
'CASK / DISK STEEL
SS304 5 1.0 293.0 END
'BORAL SHEETS
AL 6 DEN=2.6226 0.5738 293.0 END
B-10 6 DEN=2.6226 0.0450 293.0 END
B-11 6 DEN=2.6226 0.2735 293.0 END
C 6 DEN=2.6226 0.0926 293.0 END
'LEAD SHIELD
PB 7 1.0 293.0 END
'NS4FR SHIELD
B-10 8 0.0 8.553-5 293.0 END
B-11 8 0.0 3.422-4 293.0 END
AL 8 0.0 7.763-3 293.0 END
H 8 0.0 5.854-2 293.0 END
O 8 0.0 2.609-2 293.0 END
C 8 0.0 2.264-2 293.0 END
N 8 0.0 1.394-3 293.0 END
'CASK EXTERIOR WATER
H2O 9 1.0 293.0 END
'PELLET CLAD GAP WATER
H2O 10 1.0 293.0 END
END COMP
SQUAREPITCH 1.2598 0.7844 1 3 0.9144 2 0.8002 10 END
NAC-STC 26 DIRECTLY LOADED, WEST 17x17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25
READ PARAM RUN=YES PLT=NO TME=5000 GEN=803 NPG=1000 END PARAM
READ GEOM
UNIT 1
COM='FUEL PIN CELL - FOR WATER ELEVATION'
CYLINDER 1 1 0.3922 2P2.3749
CYLINDER 10 1 0.4001 2P2.3749
CYLINDER 2 1 0.4572 2P2.3749
CUBOID 3 1 4P0.6299 2P2.3749
UNIT 2
COM='GUIDE/INSTRUMENT TUBE CELL - FOR WATER ELEVATION'
CYLINDER 3 1 0.5715 2P2.3749
CYLINDER 2 1 0.6121 2P2.3749
CUBOID 3 1 4P0.6299 2P2.3749
UNIT 3
COM='FUEL PIN CELL - FOR STEEL DISK ELEVATION'
CYLINDER 1 1 0.3922 2P0.6350
CYLINDER 10 1 0.4001 2P0.6350
CYLINDER 2 1 0.4572 2P0.6350
CUBOID 3 1 4P0.6299 2P0.6350
UNIT 4
COM='GUIDE/INSTRUMENT TUBE CELL - FOR STEEL DISK ELEVATION'
CYLINDER 3 1 0.5715 2P0.6350
CYLINDER 2 1 0.6121 2P0.6350
CUBOID 3 1 4P0.6299 2P0.6350
UNIT 5
COM='FUEL PIN CELL - FOR AL DISK ELEVATION'
CYLINDER 1 1 0.3922 2P0.7938
CYLINDER 10 1 0.4001 2P0.7938
CYLINDER 2 1 0.4572 2P0.7938
CUBOID 3 1 4P0.6299 2P0.7938
UNIT 6
COM='GUIDE/INSTRUMENT TUBE CELL - FOR AL DISK ELEVATION'
CYLINDER 3 1 0.5715 2P0.7938
CYLINDER 2 1 0.6121 2P0.7938
CUBOID 3 1 4P0.6299 2P0.7938
UNIT 21
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

```
COM='ASSEMBLY - FOR WATER ELEVATION'
ARRAY 1 -10.7083 -10.7083 -2.3749
UNIT 22
COM='ASSEMBLY - FOR STEEL DISK ELEVATION'
ARRAY 2 -10.7083 -10.7083 -0.6350
UNIT 23
COM='ASSEMBLY - FOR AL DISK ELEVATION'
ARRAY 3 -10.7083 -10.7083 -0.7938
UNIT 31
COM='X-X BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P2.3749
CUBOID 4 1 2P10.3886 2P0.0951 2P2.3749
UNIT 32
COM='Y-Y BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P2.3749
CUBOID 4 1 2P0.0951 2P10.3886 2P2.3749
UNIT 33
COM='X-X BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.6350
CUBOID 4 1 2P10.3886 2P0.0951 2P0.6350
UNIT 34
COM='Y-Y BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.6350
CUBOID 4 1 2P0.0951 2P10.3886 2P0.6350
UNIT 35
COM='X-X BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.7938
CUBOID 4 1 2P10.3886 2P0.0951 2P0.7938
UNIT 36
COM='Y-Y BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.7938
CUBOID 4 1 2P0.0951 2P10.3886 2P0.7938
UNIT 40
COM='FUEL TUBE - FOR WATER ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 41
COM='FUEL TUBE - FOR WATER ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 42
COM='FUEL TUBE - FOR WATER ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 43
COM='FUEL TUBE - FOR WATER ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

```
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 44
COM='FUEL TUBE - FOR WATER ELEVATION (TL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 45
COM='FUEL TUBE - FOR WATER ELEVATION (TR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 +0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0.0 +11.4054 0.0
HOLE 31 0.0 -11.4054 0.0
HOLE 32 +11.4054 0.0 0.0
HOLE 32 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 50
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 51
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 52
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 53
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 54
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TL)'
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

```
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 55
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TR)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 +0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0.0 +11.4054 0.0
HOLE 33 0.0 -11.4054 0.0
HOLE 34 +11.4054 0.0 0.0
HOLE 34 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 60
COM='FUEL TUBE - FOR AL DISK ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0.0 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 61
COM='FUEL TUBE - FOR AL DISK ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0.0 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 62
COM='FUEL TUBE - FOR AL DISK ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 -0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 63
COM='FUEL TUBE - FOR AL DISK ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 +0.4803 -0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 64
COM='FUEL TUBE - FOR AL DISK ELEVATION (TL)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 -0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

```
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 65
COM='FUEL TUBE - FOR AL DISK ELEVATION (TR)'
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 +0.4803 +0.4803 0.0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0.0 +11.4054 0.0
HOLE 35 0.0 -11.4054 0.0
HOLE 36 +11.4054 0.0 0.0
HOLE 36 -11.4054 0.0 0.0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 70
COM='DISK OPENING - FOR WATER ELEVATION (B)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 40 0.0 -0.1810 0.0
UNIT 71
COM='DISK OPENING - FOR WATER ELEVATION (T)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 41 0.0 +0.1810 0.0
UNIT 72
COM='DISK OPENING - FOR WATER ELEVATION (BL)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 42 -0.1810 -0.1810 0.0
UNIT 73
COM='DISK OPENING - FOR WATER ELEVATION (BR)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 43 +0.1810 -0.1810 0.0
UNIT 74
COM='DISK OPENING - FOR WATER ELEVATION (TL)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 44 -0.1810 +0.1810 0.0
UNIT 75
COM='DISK OPENING - FOR WATER ELEVATION (TR)'
CUBOID 3 1 4P11.7272 2P2.3749
HOLE 45 +0.1810 +0.1810 0.0
UNIT 80
COM='DISK OPENING - FOR STEEL DISK ELEVATION (B)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 50 0.0 -0.1810 0.0
UNIT 81
COM='DISK OPENING - FOR STEEL DISK ELEVATION (T)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 51 0.0 +0.1810 0.0
UNIT 82
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BL)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 52 -0.1810 -0.1810 0.0
UNIT 83
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BR)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 53 +0.1810 -0.1810 0.0
UNIT 84
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TL)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 54 -0.1810 +0.1810 0.0
UNIT 85
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TR)'
CUBOID 3 1 4P11.7272 2P0.6350
HOLE 55 +0.1810 +0.1810 0.0
UNIT 90
COM='DISK OPENING - FOR AL DISK ELEVATION (B)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 60 0.0 -0.1810 0.0
UNIT 91
COM='DISK OPENING - FOR AL DISK ELEVATION (T)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 61 0.0 +0.1810 0.0
UNIT 92
COM='DISK OPENING - FOR AL DISK ELEVATION (BL)'
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 62 -0.1810 -0.1810 0.0
UNIT 93
COM='DISK OPENING - FOR AL DISK ELEVATION (BR)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 63 +0.1810 -0.1810 0.0
UNIT 94
COM='DISK OPENING - FOR AL DISK ELEVATION (TL)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 64 -0.1810 +0.1810 0.0
UNIT 95
COM='DISK OPENING - FOR AL DISK ELEVATION (TR)'
CUBOID 3 1 4P11.7272 2P0.7938
HOLE 65 +0.1810 +0.1810 0.0
UNIT 101
COM='BASKET STRUCTURE IN TRANSPORT CASK - WATER ELEVATION'
CYLINDER 3 1 +90.17 2P2.3749
HOLE 75 -58.9864 -40.7822 0.0
HOLE 75 -58.9864 -13.5941 0.0
HOLE 73 -58.9864 +13.5941 0.0
HOLE 73 -58.9864 +40.7822 0.0
HOLE 75 -27.1882 -67.9704 0.0
HOLE 75 -27.1882 -40.7822 0.0
HOLE 75 -27.1882 -13.5941 0.0
HOLE 73 -27.1882 +13.5941 0.0
HOLE 73 -27.1882 +40.7822 0.0
HOLE 73 -27.1882 +67.9704 0.0
HOLE 71 0.0 -67.9704 0.0
HOLE 71 0.0 -40.7822 0.0
HOLE 71 0.0 -13.5941 0.0
HOLE 70 0.0 +13.5941 0.0
HOLE 70 0.0 +40.7822 0.0
HOLE 70 0.0 +67.9704 0.0
HOLE 74 +27.1882 -67.9704 0.0
HOLE 74 +27.1882 -40.7822 0.0
HOLE 74 +27.1882 -13.5941 0.0
HOLE 72 +27.1882 +13.5941 0.0
HOLE 72 +27.1882 +40.7822 0.0
HOLE 72 +27.1882 +67.9704 0.0
HOLE 74 +58.9864 -40.7822 0.0
HOLE 74 +58.9864 -13.5941 0.0
HOLE 72 +58.9864 +13.5941 0.0
HOLE 72 +58.9864 +40.7822 0.0
CYLINDER 5 1 +93.98 2P2.3749
CYLINDER 7 1 +103.43 2P2.3749
CYLINDER 5 1 +110.11 2P2.3749
CYLINDER 9 1 +124.12 2P2.3749
CYLINDER 9 1 +124.44 2P2.3749
CYLINDER 5 1 +125.07 2P2.3749
CUBOID 9 1 4P125.07 2P2.3749
UNIT 102
COM='BASKET STRUCTURE IN TRANSPORT CASK - ST DISK ELEVATION'
CYLINDER 5 1 +89.99 2P0.6350
HOLE 85 -58.9864 -40.7822 0.0
HOLE 85 -58.9864 -13.5941 0.0
HOLE 83 -58.9864 +13.5941 0.0
HOLE 83 -58.9864 +40.7822 0.0
HOLE 85 -27.1882 -67.9704 0.0
HOLE 85 -27.1882 -40.7822 0.0
HOLE 85 -27.1882 -13.5941 0.0
HOLE 83 -27.1882 +13.5941 0.0
HOLE 83 -27.1882 +40.7822 0.0
HOLE 83 -27.1882 +67.9704 0.0
HOLE 81 0.0 -67.9704 0.0
HOLE 81 0.0 -40.7822 0.0
HOLE 81 0.0 -13.5941 0.0
HOLE 80 0.0 +13.5941 0.0
HOLE 80 0.0 +40.7822 0.0
HOLE 80 0.0 +67.9704 0.0
HOLE 84 +27.1882 -67.9704 0.0
HOLE 84 +27.1882 -40.7822 0.0
HOLE 84 +27.1882 -13.5941 0.0
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```

HOLE 82 +27.1882 +13.5941 0.0
HOLE 82 +27.1882 +40.7822 0.0
HOLE 82 +27.1882 +67.9704 0.0
HOLE 84 +58.9864 -40.7822 0.0
HOLE 84 +58.9864 -13.5941 0.0
HOLE 82 +58.9864 +13.5941 0.0
HOLE 82 +58.9864 +40.7822 0.0
CYLINDER 3 1 +90.17 2P0.6350
CYLINDER 5 1 +93.98 2P0.6350
CYLINDER 7 1 +103.43 2P0.6350
CYLINDER 5 1 +110.11 2P0.6350
CYLINDER 9 1 +124.12 2P0.6350
CYLINDER 9 1 +124.44 2P0.6350
CYLINDER 5 1 +125.07 2P0.6350
CUBOID 9 1 4P125.07 2P0.6350
UNIT 103
COM='BASKET STRUCTURE IN TRANSPORT CASK - AL DISK ELEVATION'
CYLINDER 4 1 +89.73 2P0.7938
HOLE 95 -58.9864 -40.7822 0.0
HOLE 95 -58.9864 -13.5941 0.0
HOLE 93 -58.9864 +13.5941 0.0
HOLE 93 -58.9864 +40.7822 0.0
HOLE 95 -27.1882 -67.9704 0.0
HOLE 95 -27.1882 -40.7822 0.0
HOLE 95 -27.1882 -13.5941 0.0
HOLE 93 -27.1882 +13.5941 0.0
HOLE 93 -27.1882 +40.7822 0.0
HOLE 93 -27.1882 +67.9704 0.0
HOLE 91 0.0 -67.9704 0.0
HOLE 91 0.0 -40.7822 0.0
HOLE 91 0.0 -13.5941 0.0
HOLE 90 0.0 +13.5941 0.0
HOLE 90 0.0 +40.7822 0.0
HOLE 90 0.0 +67.9704 0.0
HOLE 94 +27.1882 -67.9704 0.0
HOLE 94 +27.1882 -40.7822 0.0
HOLE 94 +27.1882 -13.5941 0.0
HOLE 92 +27.1882 +13.5941 0.0
HOLE 92 +27.1882 +40.7822 0.0
HOLE 92 +27.1882 +67.9704 0.0
HOLE 94 +58.9864 -40.7822 0.0
HOLE 94 +58.9864 -13.5941 0.0
HOLE 92 +58.9864 +13.5941 0.0
HOLE 92 +58.9864 +40.7822 0.0
CYLINDER 3 1 +90.17 2P0.7938
CYLINDER 5 1 +93.98 2P0.7938
CYLINDER 7 1 +103.43 2P0.7938
CYLINDER 5 1 +110.11 2P0.7938
CYLINDER 9 1 +124.12 2P0.7938
CYLINDER 9 1 +124.44 2P0.7938
CYLINDER 5 1 +125.07 2P0.7938
CUBOID 9 1 4P125.07 2P0.7938
GLOBAL UNIT 104
COM='DISK SLICE STACK'
ARRAY 4 -125.07 -125.07 0.0
CUBOID 9 1 4P125.08 12.3573 0.0
END GEOM
READ ARRAY
ARA=1 NUX=17 NUY=17 NUZ=1 FILL
34R1
5R1 2 2R1 2 2R1 2 5R1
3R1 2 9R1 2 3R1
17R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
17R1
3R1 2 9R1 2 3R1
5R1 2 2R1 2 2R1 2 5R1
34R1

```


Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

```
END FILL
ARA=2 NUX=17 NUY=17 NUZ=1 FILL
      34R3
      5R3 4 2R3 4 2R3 4 5R3
      3R3 4 9R3 4 3R3
      17R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3 4 2R3
      34R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3 4 2R3
      34R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3 4 2R3
      17R3
      3R3 4 9R3 4 3R3
      5R3 4 2R3 4 2R3 4 5R3
      34R3

END FILL
ARA=3 NUX=17 NUY=17 NUZ=1 FILL
      34R5
      5R5 6 2R5 6 2R5 6 5R5
      3R5 6 9R5 6 3R5
      17R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5 6 2R5
      34R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5 6 2R5
      34R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5 6 2R5
      17R5
      3R5 6 9R5 6 3R5
      5R5 6 2R5 6 2R5 6 5R5
      34R5

END FILL
ARA=4 NUX=1 NUY=1 NUZ=4 FILL 101 102 101 103 END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIRROR END BOUNDS
READ PLOT
TTL='XY SLICE OF CASK - ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-120.0 YUL=120.0 ZUL=5.5 XLR=120.0 YLR=-120.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='XY SLICE CASK CENTER AREA ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-27.0 YUL=27.0 ZUL=5.5 XLR=27.0 YLR=-27.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
END PLOT
END DATA
```

[illegible]

***** PROBLEM PARAMETERS *****

**** PROBLEM COMPOSITION DESCRIPTION ****

```
' CLAD
      END
```

6.7-36

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
            40302      1.00 ATOM/MOLECULE

'H2O CASK INTERIOR
END

SC H2O          STANDARD COMPOSITION
MX              3 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          0.9982 THEORETICAL DENSITY
NEL           2 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
            1001      2.00 ATOMS/MOLECULE
            8016      1.00 ATOM/MOLECULE

'AL DISK
END

SC AL          STANDARD COMPOSITION
MX              4 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          2.7020 THEORETICAL DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
            13027      1.00 ATOM/MOLECULE

'CASK / DISK STEEL
END

SC SS304        STANDARD COMPOSITION
MX              5 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          7.9200 THEORETICAL DENSITY
NEL           4 NO. ELEMENTS
ICP           0 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
            24304      19.000 WT%
            25055       2.000 WT%
            26304      69.500 WT%
            28304       9.500 WT%

'BORAL SHEETS
END

SC AL          STANDARD COMPOSITION
MX              6 MIXTURE NO.
VF            0.5738 VOLUME FRACTION
ROTH          2.6226 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
            13027      1.00 ATOM/MOLECULE
END

SC B-10         STANDARD COMPOSITION
MX              6 MIXTURE NO.
VF            0.0450 VOLUME FRACTION
ROTH          2.6226 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
            5010      1.00 ATOM/MOLECULE
END

SC B-11         STANDARD COMPOSITION
MX              6 MIXTURE NO.
VF            0.2735 VOLUME FRACTION
ROTH          2.6226 SPECIFIED DENSITY
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              5011      1.00 ATOM/MOLECULE
END

SC C          STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.0926 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              6012      1.00 ATOM/MOLECULE

'LEAD SHIELD
END

SC PB          STANDARD COMPOSITION
MX           7 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        11.3440 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              82000     1.00 ATOM/MOLECULE

'NS4FR SHIELD
END

SC B-10        STANDARD COMPOSITION
MX            8 MIXTURE NO.
DEN          8.5530E-05 ATOMIC DENSITY
ROTH         1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              5010      1.00 ATOM/MOLECULE
END

SC B-11        STANDARD COMPOSITION
MX            8 MIXTURE NO.
DEN          3.4220E-04 ATOMIC DENSITY
ROTH         1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              5011      1.00 ATOM/MOLECULE
END

SC AL          STANDARD COMPOSITION
MX            8 MIXTURE NO.
DEN          7.7630E-03 ATOMIC DENSITY
ROTH         2.7020 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              13027     1.00 ATOM/MOLECULE
END

SC H           STANDARD COMPOSITION
MX            8 MIXTURE NO.
DEN          5.8540E-02 ATOMIC DENSITY
ROTH         1.0000 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
              1001      1.00 ATOM/MOLECULE
END

SC O           STANDARD COMPOSITION
MX            8 MIXTURE NO.
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```
DEN 2.6090E-02 ATOMIC DENSITY
ROTH 1.0000 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      8016 1.00 ATOM/MOLECULE
END

SC C STANDARD COMPOSITION
MX 8 MIXTURE NO.
DEN 2.2640E-02 ATOMIC DENSITY
ROTH 2.1000 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      6012 1.00 ATOM/MOLECULE
END

SC N STANDARD COMPOSITION
MX 8 MIXTURE NO.
DEN 1.3940E-03 ATOMIC DENSITY
ROTH 1.0000 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      7014 1.00 ATOM/MOLECULE

'CASK EXTERIOR WATER
END

SC H2O STANDARD COMPOSITION
MX 9 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 0.9982 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      1001 2.00 ATOMS/MOLECULE
      8016 1.00 ATOM/MOLECULE

'PELLET CLAD GAP WATER
END

SC H2O STANDARD COMPOSITION
MX 10 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 0.9982 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      1001 2.00 ATOMS/MOLECULE
      8016 1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH 1.2598 CM CENTER TO CENTER SPACING
FUELOD 0.7844 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL 1 MIXTURE NO. OF FUEL
MMOD 3 MIXTURE NO. OF MODERATOR
CLADOD 0.9144 CM CLAD OUTER DIAMETER
MCLAD 2 MIXTURE NO. OF CLAD
GAPOD 0.8002 CM GAP OUTER DIAMETER
MGAP 10 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```
*****  
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
***** PROGRAM: O0O002 *****  
***** CREATION DATE: 09-28-95 *****  
***** VOLUME: ENG *****  
***** LIBRARY: G:\SCALE43\EXE *****  
***** PRODUCTION CODE: NITAWL *****  
***** VERSION: 3.0 *****  
***** JOBNAME: SCALE-PC *****  
***** DATE OF EXECUTION: 02/03/98 *****  
***** TIME OF EXECUTION: 07:42:53 *****  
*****  
*****  
*****  
  
-1Q ARRAY HAS 1 ENTRIES.  
  
0Q ARRAY HAS 9 ENTRIES.  
  
1Q ARRAY HAS 12 ENTRIES.  
  
SELECT 27 NUCLIDES FROM THE MASTER LIBRARY ON LOGICAL 1  
.0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL 2  
0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL 3  
TO CREATE THE NEW WORKING LIBRARY ON LOGICAL 4  
  
4 RESONANCE CALCULATIONS HAVE BEEN REQUESTED  
-1 OUTPUT OPTION FOR AMPX FORMATTED CROSS SECTION DATA  
2001 MAXIMUM NUMBER OF RESONANCE MESH INTERVALS  
2 ORDER OF RESONANCE LEVEL PROCESSING  
  
THE STORAGE ALLOCATED FOR THIS CASE IS 100000 WORDS  
  
2Q ARRAY HAS 27 ENTRIES.  
  
3Q ARRAY HAS 60 ENTRIES.  
  
4Q ARRAY HAS 27 ENTRIES.  
  
GENERAL INFORMATION CONCERNING CROSS SECTION LIBRARY  
TAPE IDENTIFICATION NUMBER 4321  
NUMBER OF NUCLIDES ON TAPE 27  
NUMBER OF NEUTRON ENERGY GROUPS 27  
FIRST THERMAL NEUTRON ENERGY GROUP 15  
NUMBER OF GAMMA ENERGY GROUPS 0  
  
DIRECT ACCESS UNIT NUMBER 9 REQUIRES 117 BLOCKS OF LENGTH 1680 WORDS  
XSDRN TAPE 4321
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY					
BASED ON ENDF-B VERSION 4 DATA					
COMPILED FOR NRC 1/27/89					
LAST UPDATED					
L.M.PETRIE - ORNL					
08/12/94					
NUCLIDES FROM XSDRN TAPE					
1	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	
2	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	
3	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	
4	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	10001001	
5	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010	
6	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010	
7	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011	
8	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011	
9	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012	
10	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012	
11	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014	
12	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016	
13	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	3008016	
14	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	8008016	
15	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	9008016	
16	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	10008016	
17	AL-27 1193 218 GP 040375 (5)		UPDATED 08/12/94	4013027	
18	AL-27 1193 218 GP 040375 (5)		UPDATED 08/12/94	6013027	
19	AL-27 1193 218 GP 040375 (5)		UPDATED 08/12/94	8013027	
20	CR 1191 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)		UPDATED 08/12/94	5024304	
21	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94	5025055	
22	FE 1192 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)		UPDATED 08/12/94	5026304	
23	NI 1190 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)		UPDATED 08/12/94	5028304	
24	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94	2040302	
25	PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94	7082000	
26	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94	1092235	
27	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94	1092238	
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	10001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=		293.00

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 3008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 8008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 9008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 10008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 4013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 6013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 8013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94 5024304	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94 5025055	TEMPERATURE=	293.00
GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00				
RESONANCE DATA FOR THIS NUCLIDE				
MASS NUMBER (A)	=	54.466	TEMPERATURE (KELVIN)	= 293.000
POTENTIAL SCATTER SIGMA	=	2.590	LUMPED NUCLEAR DENSITY	= 1.7363295E-03
SPIN FACTOR (G)	=	14.448	LUMP DIMENSION (A-BAR)	= 0.0000000E+00
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	= 0.0000000E+00
THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.				
MASS OF MODERATOR-1	=	55.845	SIGMA (PER ABSORBER ATOM)	= 3.4663022E+02
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.				
MASS OF MODERATOR-2	=	55.925	SIGMA (PER ABSORBER ATOM)	= 1.2557598E+02
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.				
THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.				
VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000				
GROUP	RES ABS	RES FISS	RES SCAT	
8	-5.518788E-04	0.000000E+00	-3.944190E-01	
9	-2.797993E-03	0.000000E+00	-2.293471E+00	
10	-3.291452E-01	0.000000E+00	-3.820862E+01	
11	-2.680562E+00	0.000000E+00	-1.159996E+02	
EXCESS RESONANCE INTEGRALS				
RESOLVED				
ABSORPTION	3.33719E+00			
FISSION	0.00000E+00			
			PROCESS NUMBER 1007 IS AT	TEMPERATURE= 293.00
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94 5026304	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94 5028304	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

ZIRCALLOY ENDF/B-IV MAT 1284 UPDATED 08/12/94 2040302 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 90.436 TEMPERATURE(KELVIN) = 293.000

POTENTIAL SCATTER SIGMA = 6.385 LUMPED NUCLEAR DENSITY = 4.3307818E-02

SPIN FACTOR (G) = 1.079 LUMP DIMENSION (A-BAR) = 4.5719999E-01

INNER RADIUS = 4.0009999E-01 DANCORF CORRECTION (C) = 4.4366416E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-3.284353E-04	0.000000E+00	-2.612845E-01
9	-2.311753E-02	0.000000E+00	-9.524279E-01
10	-2.325708E-02	0.000000E+00	-5.342349E-01
11	-8.027643E-02	0.000000E+00	-3.840209E-01

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 5.62305E-01

FISSION 0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

PB 1288 218NGP 042375 P-3 293K UPDATED 08/12/94 7082000 TEMPERATURE= 293.00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-235 ENDF/B-IV MAT 1261 UPDATED 08/12/94 1092235 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 233.025 TEMPERATURE(KELVIN) = 293.000

POTENTIAL SCATTER SIGMA = 11.500 LUMPED NUCLEAR DENSITY = 9.8766887E-04

SPIN FACTOR (G) = 15171.100 LUMP DIMENSION (A-BAR) = 3.9219999E-01

INNER RADIUS = 0.0000000E+00 DANCORF CORRECTION (C) = 2.3525730E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA(PER ABSORBER ATOM)= 1.8285364E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 238.051 SIGMA(PER ABSORBER ATOM)= 2.7760712E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
12	-3.904891E+00	-2.400683E+00	-9.365656E-02
13	-1.223861E+01	-5.975270E+00	-2.623999E-01
14	-9.154386E+00	-5.417213E+00	-6.158343E-02
15	-5.171851E-04	-3.930799E-04	4.654119E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 1.98617E+02

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

FISSION 1.19162E+02
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
URANIUM-238 ENDF/B-IV MAT 1262 UPDATED 08/12/94 1092238 TEMPERATURE= 293.00
RESONANCE DATA FOR THIS NUCLIDE
MASS NUMBER (A) = 236.006 TEMPERATURE (KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 10.599 LUMPED NUCLEAR DENSITY = 2.2243705E-02
SPIN FACTOR (G) = 656.527 LUMP DIMENSION (A-BAR) = 3.9219999E-01
INNER RADIUS = 0.0000000E+00 DANCORFF CORRECTION (C) = 2.3525730E-01
THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
MASS OF MODERATOR-1 = 15.991 SIGMA (PER ABSORBER ATOM) = 8.1190996E+00
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
MASS OF MODERATOR-2 = 235.044 SIGMA (PER ABSORBER ATOM) = 5.2849644E-01
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.
VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000
GROUP RES ABS RES FISS RES SCAT
9 -4.655258E-02 0.000000E+00 -4.609913E-01
10 -1.139702E+00 -2.338724E-05 -6.895151E+00
11 -9.892670E+00 0.000000E+00 -2.712805E+01
12 -4.299526E+01 0.000000E+00 -4.994605E+01
13 -5.378721E+01 0.000000E+00 -1.765308E+01
14 -1.041276E+02 0.000000E+00 -6.054823E+00
15 -8.495994E-07 0.000000E+00 1.647857E-06
EXCESS RESONANCE INTEGRALS
RESOLVED
ABSORPTION 1.88041E+01
FISSION 4.93936E-04
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
THIS XSDRN WORKING TAPE WAS CREATED 02/03/98 AT 07:42:53
THE TITLE OF THE PARENT CASE IS AS FOLLOWS
SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
BASED ON ENDF-B VERSION 4 DATA
COMPILED FOR NRC 1/27/89
TAPE ID 4321 NUMBER OF NUCLIDES 27
NUMBER OF NEUTRON GROUPS 27 NUMBER OF GAMMA GROUPS 0
FIRST THERMAL GROUP 15 LOGICAL UNIT 4
TABLE OF CONTENTS
HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 ID 3001001
HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 ID 8001001
HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 ID 9001001
HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 ID 10001001
B-10 1273 218NGP 042375 P-3 293K UPDATED 08/12/94 ID 6005010
B-10 1273 218NGP 042375 P-3 293K UPDATED 08/12/94 ID 8005010
BORON-11 ENDF/B-IV MAT 1160 UPDATED 08/12/94 ID 6005011
BORON-11 ENDF/B-IV MAT 1160 UPDATED 08/12/94 ID 8005011
CARBON-12 ENDF/B-IV MAT 1274/THRM1065 UPDATED 08/12/94 ID 6006012
CARBON-12 ENDF/B-IV MAT 1274/THRM1065 UPDATED 08/12/94 ID 8006012
NITROGEN-14 ENDF/B-IV MAT 1275 UPDATED 08/12/94 ID 8007014
OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 ID 1008016
OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 ID 3008016
OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 ID 8008016
OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 ID 9008016
OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 ID 1008016
AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 ID 4013027
AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 ID 6013027

AL-27 1193 218 GP 040375(5)	UPDATED	08/12/94	ID	8013027
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'	UPDATED	08/12/94	ID	5024304
MANGANESE-55 ENDF/B-IV MAT 1197	UPDATED	08/12/94	ID	5025055
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'	UPDATED	08/12/94	ID	5026304
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'	UPDATED	08/12/94	ID	5028304
ZIRCALLOY ENDF/B-IV MAT 1284	UPDATED	08/12/94	ID	2040302
PB 1288 218NGP 042375 P-3 293K	UPDATED	08/12/94	ID	7082000
URANIUM-235 ENDF/B-IV MAT 1261	UPDATED	08/12/94	ID	1092235
URANIUM-238 ENDF/B-IV MAT 1262	UPDATED	08/12/94	ID	1092238

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
*****  
*****  
***** PROGRAM: O00009 *****  
*****  
***** CREATION DATE: 03-08-96 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
***** PRODUCTION CODE: KENOVA *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 02/03/98 *****  
*****  
***** TIME OF EXECUTION: 07:43:08 *****  
*****  
*****
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```

*****
***                                     *****
***                                     NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OPA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25
***                                     *****
***                                     NUMERIC PARAMETERS
***                                     *****
***
***      TME      MAXIMUM PROBLEM TIME (MIN)      *****
***
***      TBA      TIME PER GENERATION (MIN)      0.50
***
***      GEN      NUMBER OF GENERATIONS      803
***
***      NPG      NUMBER PER GENERATION      1000
***
***      NSK      NUMBER OF GENERATIONS TO BE SKIPPED      3
***
***      BEG      BEGINNING GENERATION NUMBER      1
***
***      RES      GENERATIONS BETWEEN CHECKPOINTS      0
***
***      X1D      NUMBER OF EXTRA 1-D CROSS SECTIONS      1
***
***      NBK      NEUTRON BANK SIZE      1025
***
***      XNB      EXTRA POSITIONS IN NEUTRON BANK      0
***
***      NFB      FISSION BANK SIZE      1000
***
***      XFB      EXTRA POSITIONS IN FISSION BANK      0
***
***      WTA      DEFAULT VALUE OF WEIGHT AVERAGE      0.5000
***
***      WTH      WEIGHT HIGH FOR SPLITTING      3.0000
***
***      WTL      WEIGHT LOW FOR RUSSIAN ROULETTE      0.3333
***
***      RND      STARTING RANDOM NUMBER      BB827100001
***
***      NB8      NUMBER OF D.A. BLOCKS ON UNIT 8      200
***
***      NL8      LENGTH OF D.A. BLOCKS ON UNIT 8      512
***
***      ADJ      MODE OF CALCULATION      FORWARD
***
***      INPUT DATA WRITTEN ON RESTART UNIT      NO
***
***      BINARY DATA INTERFACE      YES
***
*****

```

***** LOGICAL PARAMETERS *****						
RUN	EXECUTE PROBLEM AFTER CHECKING DATA	YES	PLT	PLOT PICTURE MAP(S)		NO
FLX	COMPUTE FLUX	NO	FDN	COMPUTE FISSION DENSITIES		NO
SMU	COMPUTE AVG UNIT SELF-MULTIPLICATION	NO	NUB	COMPUTE NU-BAR & AVG FISSION GROUP		YES
MKU	COMPUTE MATRIX K-EFF BY UNIT NUMBER	NO	MKP	COMPUTE MATRIX K-EFF BY UNIT LOCATION		NO
CKU	COMPUTE COFACTOR K-EFF BY UNIT NUMBER	NO	CKP	COMPUTE COFACTOR K-EFF BY UNIT LOCATION		NO
FMU	PRINT FISSION PROD MATRIX BY UNIT NUMBER	NO	FMP	PRINT FISSION PROD MATRIX BY UNIT LOCATION		NO
MKH	COMPUTE MATRIX K-EFF BY HOLE NUMBER	NO	MKA	COMPUTE MATRIX K-EFF BY ARRAY NUMBER		NO
CKH	COMPUTE COFACTOR K-EFF BY HOLE NUMBER	NO	CKA	COMPUTE COFACTOR K-EFF BY ARRAY NUMBER		NO
FMH	PRINT FISSION PROD MATRIX BY HOLE NUMBER	NO	FMA	PRINT FISSION PROD MATRIX BY ARRAY NUMBER		NO
HHL	COLLECT MATRIX BY HIGHEST HOLE LEVEL	NO	HAL	COLLECT MATRIX BY HIGHEST ARRAY LEVEL		NO
AMX	PRINT ALL MIXED CROSS SECTIONS	NO	FAR	PRINT FIS. AND ABS. BY REGION		NO
XS1	PRINT 1-D MIXTURE X-SECTIONS	NO	GAS	PRINT FAR BY GROUP		NO
XS2	PRINT 2-D MIXTURE X-SECTIONS	NO	PAX	PRINT XSEC-ALBEDO CORRELATION TABLES		NO
XAP	PRINT MIXTURE ANGLES & PROBABILITIES	NO	PWT	PRINT WEIGHT AVERAGE ARRAY		NO
PKI	PRINT FISSION SPECTRUM	NO	PGM	PRINT INPUT GEOMETRY		NO
P1D	PRINT EXTRA 1-D CROSS SECTIONS	NO	BUG	PRINT DEBUG INFORMATION		NO
			TRK	PRINT TRACKING INFORMATION		NO

PARAMETER INPUT COMPLETED

MAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD = 3.0E-05

6.7-47

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

MIXTURE =	3	DENSITY(G/CC) =	0.99817						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
3001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002		UPDATED	
08/12/94									
3008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276		UPDATED	
08/12/94									
MIXTURE =	4	DENSITY(G/CC) =	2.7020						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
4013027	6.03066E-02	1.00000E+00	13027	26.9818	AL-27 1193 218 GP 040375(5)			UPDATED	
08/12/94									
MIXTURE =	5	DENSITY(G/CC) =	7.9200						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
5024304	1.74286E-02	1.90000E-01	24000	51.9957	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)			UPDATED	
08/12/94									
5025055	1.73633E-03	1.99999E-02	25055	54.9379	MANGANESE-55	ENDF/B-IV MAT 1197		UPDATED	
08/12/94									
5026304	5.93579E-02	6.95000E-01	26000	55.8447	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)			UPDATED	
08/12/94									
5028304	7.72070E-03	9.50001E-02	28000	58.6872	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)			UPDATED	
08/12/94									
MIXTURE =	6	DENSITY(G/CC) =	2.5830						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
6005010	7.09799E-03	4.56901E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K			UPDATED	
08/12/94									
6005011	3.92356E-02	2.77698E-01	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160		UPDATED	
08/12/94									
6006012	1.21874E-02	9.40196E-02	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065		UPDATED	
08/12/94									
6013027	3.35871E-02	5.82592E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)			UPDATED	
08/12/94									
MIXTURE =	7	DENSITY(G/CC) =	11.344						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K			UPDATED	
08/12/94									
MIXTURE =	8	DENSITY(G/CC) =	1.6298						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
8001001	5.85400E-02	6.01023E-02	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002		UPDATED	
08/12/94									
8005010	8.55300E-05	8.72589E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K			UPDATED	
08/12/94									
8005011	3.42200E-04	3.83863E-03	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160		UPDATED	
08/12/94									
8006012	2.26400E-02	2.76813E-01	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065		UPDATED	
08/12/94									
8007014	1.39400E-03	1.98893E-02	7014	14.0033	NITROGEN-14	ENDF/B-IV MAT 1275		UPDATED	
08/12/94									
8008016	2.60900E-02	4.25068E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276		UPDATED	
08/12/94									
8013027	7.76300E-03	2.13416E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)			UPDATED	
08/12/94									
MIXTURE =	9	DENSITY(G/CC) =	0.99817						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
9001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002		UPDATED	
08/12/94									
9008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276		UPDATED	
08/12/94									
MIXTURE =	10	DENSITY(G/CC) =	0.99817						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE				
10001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002		UPDATED	
08/12/94									
10008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276		UPDATED	
08/12/94									

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

3001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
8001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
9001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
10001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
6005010	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
8005010	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
6005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
8005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
6006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
8006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
8007014	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94
1008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
3008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
8008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
9008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
10008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94
4013027	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
6013027	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
8013027	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94
5024304	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
5025055	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94
5026304	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
5028304	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94
2040302	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94
7082000	PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94
1092235	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94
1092238	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94

KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 3 WERE CORRECTED FOR BAD MOMENTS.

KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 9 WERE CORRECTED FOR BAD MOMENTS.

KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 10 WERE CORRECTED FOR BAD MOMENTS.

..... 0 IO'S WERE USED MIXING CROSS-SECTIONS

1-D CROSS SECTION ARRAY ID NUMBERS

1 2002 1452 27 18 1018

..... 0 IO'S WERE USED PREPARING THE CROSS SECTIONS

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

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*****
***
***      NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25      ***
***
*****
***** ADDITIONAL INFORMATION *****
*****
*** NUMBER OF ENERGY GROUPS          27      USE LATTICE GEOMETRY          YES ***
*** NO. OF FISSION SPECTRUM SOURCE GROUP 1      GLOBAL ARRAY NUMBER          4 ***
*** NO. OF SCATTERING ANGLES IN XSECS    2      NUMBER OF UNITS IN THE GLOBAL X DIR.    1 ***
*** ENTRIES/NEUTRON IN THE NEUTRON BANK 25      NUMBER OF UNITS IN THE GLOBAL Y DIR.    1 ***
*** ENTRIES/NEUTRON IN THE FISSION BANK 18      NUMBER OF UNITS IN THE GLOBAL Z DIR.    4 ***
*** NUMBER OF MIXTURES USED              9      USE A GLOBAL REFLECTOR          YES ***
*** NUMBER OF BIAS ID'S USED             1      USE NESTED HOLES                YES ***
*** NUMBER OF DIFFERENTIAL ALBEDOS USED   0      NUMBER OF HOLES                186 ***
*** TOTAL INPUT GEOMETRY REGIONS         154     MAXIMUM HOLE NESTING LEVEL        3 ***
*** NUMBER OF GEOMETRY REGIONS USED       154     USE NESTED ARRAYS                YES ***
*** LARGEST GEOMETRY UNIT NUMBER          104     NUMBER OF ARRAYS USED            4 ***
*** LARGEST ARRAY NUMBER                  4      MAXIMUM ARRAY NESTING LEVEL        2 ***
***
*** +X BOUNDARY CONDITION      MIRROR      -X BOUNDARY CONDITION      MIRROR ***
*** +Y BOUNDARY CONDITION      MIRROR      -Y BOUNDARY CONDITION      MIRROR ***
*** +Z BOUNDARY CONDITION      PER         -Z BOUNDARY CONDITION      PER ***
*****

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

      GENERATION      ELAPSED TIME      AVERAGE      AVG K-EFF      MATRIX      MATRIX K-EFF
      GENERATION      K-EFFECTIVE      MINUTES      K-EFFECTIVE      DEVIATION      K-EFFECTIVE      DEVIATION
KENO MESSAGE NUMBER K5-132  WARNING...ONLY  915 INDEPENDENT FISSION POINTS WERE GENERATED
1      8.45028E-01      1.85000E-01      1.00000E+00      0.00000E+00      0.00000E+00      0.00000E+00
KENO MESSAGE NUMBER K5-132  WARNING...ONLY  954 INDEPENDENT FISSION POINTS WERE GENERATED
2      8.60486E-01      2.54500E-01      1.00000E+00      0.00000E+00      0.00000E+00      0.00000E+00
KENO MESSAGE NUMBER K5-132  WARNING...ONLY  962 INDEPENDENT FISSION POINTS WERE GENERATED
3      8.78953E-01      3.14000E-01      8.78953E-01      0.00000E+00      0.00000E+00      0.00000E+00
4      8.80614E-01      3.97333E-01      8.79784E-01      8.30412E-04      0.00000E+00      0.00000E+00
5      9.02418E-01      4.62333E-01      8.87328E-01      7.55995E-03      0.00000E+00      0.00000E+00
6      8.56431E-01      5.31000E-01      8.79604E-01      9.39371E-03      0.00000E+00      0.00000E+00
7      9.04664E-01      5.96000E-01      8.84616E-01      8.83544E-03      0.00000E+00      0.00000E+00
8      8.52169E-01      6.58167E-01      8.79208E-01      9.01598E-03      0.00000E+00      0.00000E+00
9      8.93579E-01      7.48833E-01      8.81261E-01      7.89162E-03      0.00000E+00      0.00000E+00
10     9.14812E-01      8.08333E-01      8.85455E-01      8.01854E-03      0.00000E+00      0.00000E+00
11     8.85195E-01      8.77833E-01      8.85426E-01      7.07174E-03      0.00000E+00      0.00000E+00
12     8.74378E-01      9.50167E-01      8.84321E-01      6.42092E-03      0.00000E+00      0.00000E+00
13     9.20378E-01      1.03900E+00      8.87599E-01      6.66906E-03      0.00000E+00      0.00000E+00
14     9.05397E-01      1.11050E+00      8.89082E-01      6.26605E-03      0.00000E+00      0.00000E+00
15     9.37424E-01      1.17900E+00      8.92801E-01      6.85936E-03      0.00000E+00      0.00000E+00
16     9.19052E-01      1.28717E+00      8.94676E-01      6.62157E-03      0.00000E+00      0.00000E+00
17     9.02743E-01      1.36583E+00      8.95214E-01      6.18776E-03      0.00000E+00      0.00000E+00
18     8.71399E-01      1.43533E+00      8.93725E-01      5.97643E-03      0.00000E+00      0.00000E+00
19     9.15408E-01      1.51050E+00      8.95001E-01      5.75693E-03      0.00000E+00      0.00000E+00
20     8.69583E-01      1.61200E+00      8.93589E-01      5.60837E-03      0.00000E+00      0.00000E+00
21     9.08974E-01      1.75400E+00      8.94398E-01      5.36643E-03      0.00000E+00      0.00000E+00
22     9.43581E-01      1.88217E+00      8.96858E-01      5.65385E-03      0.00000E+00      0.00000E+00
23     8.98740E-01      1.95083E+00      8.96947E-01      5.37863E-03      0.00000E+00      0.00000E+00

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Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality
Analysis (continued)

24	9.41417E-01	2.00750E+00	8.98969E-01	5.51232E-03	0.00000E+00	0.00000E+00
25	8.95500E-01	2.09183E+00	8.98818E-01	5.26936E-03	0.00000E+00	0.00000E+00
26	9.25465E-01	2.20167E+00	8.99928E-01	5.16576E-03	0.00000E+00	0.00000E+00
27	8.96219E-01	2.39483E+00	8.99780E-01	4.95704E-03	0.00000E+00	0.00000E+00
28	8.98023E-01	2.44700E+00	8.99712E-01	4.76305E-03	0.00000E+00	0.00000E+00
29	9.22820E-01	2.75917E+00	9.00568E-01	4.66247E-03	0.00000E+00	0.00000E+00
30	9.04949E-01	3.02733E+00	9.00725E-01	4.49559E-03	0.00000E+00	0.00000E+00
31	8.82567E-01	3.26983E+00	9.00098E-01	4.38276E-03	0.00000E+00	0.00000E+00
32	9.24196E-01	3.50617E+00	9.00902E-01	4.30966E-03	0.00000E+00	0.00000E+00
33	8.99021E-01	3.60033E+00	9.00841E-01	4.16877E-03	0.00000E+00	0.00000E+00
34	9.05256E-01	3.76517E+00	9.00979E-01	4.03875E-03	0.00000E+00	0.00000E+00
35	9.11266E-01	3.81733E+00	9.01291E-01	3.92684E-03	0.00000E+00	0.00000E+00
36	8.90096E-01	3.86867E+00	9.00961E-01	3.82380E-03	0.00000E+00	0.00000E+00
37	9.34055E-01	3.97483E+00	9.01907E-01	3.83144E-03	0.00000E+00	0.00000E+00
38	9.09443E-01	4.16800E+00	9.02116E-01	3.72937E-03	0.00000E+00	0.00000E+00
39	9.20308E-01	4.29067E+00	9.02608E-01	3.66035E-03	0.00000E+00	0.00000E+00
40	9.60828E-01	4.34467E+00	9.04140E-01	3.87818E-03	0.00000E+00	0.00000E+00
41	8.92753E-01	4.40050E+00	9.03848E-01	3.78870E-03	0.00000E+00	0.00000E+00
42	9.30630E-01	4.46083E+00	9.04518E-01	3.75298E-03	0.00000E+00	0.00000E+00
43	9.30950E-01	4.51667E+00	9.05162E-01	3.71664E-03	0.00000E+00	0.00000E+00
44	8.89295E-01	4.57067E+00	9.04784E-01	3.64669E-03	0.00000E+00	0.00000E+00
.						
766	9.07771E-01	5.74732E+01	9.18980E-01	8.71601E-04	0.00000E+00	0.00000E+00
767	9.00615E-01	5.75510E+01	9.18956E-01	8.70792E-04	0.00000E+00	0.00000E+00
768	9.62929E-01	5.77148E+01	9.19013E-01	8.71547E-04	0.00000E+00	0.00000E+00
769	9.27617E-01	5.78750E+01	9.19025E-01	8.70483E-04	0.00000E+00	0.00000E+00
770	8.98795E-01	5.79968E+01	9.18998E-01	8.69747E-04	0.00000E+00	0.00000E+00
771	9.26927E-01	5.80892E+01	9.19008E-01	8.68677E-04	0.00000E+00	0.00000E+00
772	9.08778E-01	5.82338E+01	9.18995E-01	8.67650E-04	0.00000E+00	0.00000E+00
773	9.32480E-01	5.82852E+01	9.19013E-01	8.66700E-04	0.00000E+00	0.00000E+00
774	9.13695E-01	5.83363E+01	9.19006E-01	8.65604E-04	0.00000E+00	0.00000E+00
775	9.59137E-01	5.83858E+01	9.19058E-01	8.66041E-04	0.00000E+00	0.00000E+00
776	9.24788E-01	5.84362E+01	9.19065E-01	8.64953E-04	0.00000E+00	0.00000E+00
777	8.53001E-01	5.84875E+01	9.18980E-01	8.68032E-04	0.00000E+00	0.00000E+00
778	9.37161E-01	5.85360E+01	9.19003E-01	8.67229E-04	0.00000E+00	0.00000E+00
779	9.24118E-01	5.85845E+01	9.19010E-01	8.66138E-04	0.00000E+00	0.00000E+00
780	8.84378E-01	5.86367E+01	9.18965E-01	8.66168E-04	0.00000E+00	0.00000E+00
781	9.13175E-01	5.86852E+01	9.18958E-01	8.65087E-04	0.00000E+00	0.00000E+00
782	9.33630E-01	5.87382E+01	9.18977E-01	8.64182E-04	0.00000E+00	0.00000E+00
783	9.44099E-01	5.87977E+01	9.19009E-01	8.63674E-04	0.00000E+00	0.00000E+00
784	9.75015E-01	5.88692E+01	9.19081E-01	8.65537E-04	0.00000E+00	0.00000E+00
785	9.23788E-01	5.89185E+01	9.19087E-01	8.64452E-04	0.00000E+00	0.00000E+00
786	8.91437E-01	5.89698E+01	9.19051E-01	8.64069E-04	0.00000E+00	0.00000E+00
787	8.86729E-01	5.90230E+01	9.19010E-01	8.63949E-04	0.00000E+00	0.00000E+00
788	8.94312E-01	5.90770E+01	9.18979E-01	8.63421E-04	0.00000E+00	0.00000E+00
789	9.30979E-01	5.91263E+01	9.18994E-01	8.62458E-04	0.00000E+00	0.00000E+00
790	8.91876E-01	5.92070E+01	9.18960E-01	8.62050E-04	0.00000E+00	0.00000E+00
791	9.45892E-01	5.92582E+01	9.18994E-01	8.61634E-04	0.00000E+00	0.00000E+00
792	8.77510E-01	5.93077E+01	9.18941E-01	8.62143E-04	0.00000E+00	0.00000E+00
793	9.00762E-01	5.93580E+01	9.18918E-01	8.61359E-04	0.00000E+00	0.00000E+00
794	9.33553E-01	5.94093E+01	9.18937E-01	8.60469E-04	0.00000E+00	0.00000E+00
795	9.40924E-01	5.94597E+01	9.18964E-01	8.59830E-04	0.00000E+00	0.00000E+00
796	9.59407E-01	5.95337E+01	9.19015E-01	8.60256E-04	0.00000E+00	0.00000E+00
797	9.00746E-01	5.95850E+01	9.18992E-01	8.59481E-04	0.00000E+00	0.00000E+00
798	9.08014E-01	5.96573E+01	9.18979E-01	8.58511E-04	0.00000E+00	0.00000E+00
799	9.49942E-01	5.97160E+01	9.19017E-01	8.58313E-04	0.00000E+00	0.00000E+00
800	9.20550E-01	5.97690E+01	9.19019E-01	8.57239E-04	0.00000E+00	0.00000E+00
801	9.01510E-01	5.98387E+01	9.18997E-01	8.56445E-04	0.00000E+00	0.00000E+00
802	9.21593E-01	5.98880E+01	9.19001E-01	8.55380E-04	0.00000E+00	0.00000E+00
803	8.95358E-01	5.99375E+01	9.18971E-01	8.54822E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123 EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.
NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

LIFETIME = 4.03233E-05 + OR - 8.15400E-08 GENERATION TIME = 3.02951E-05 + OR - 4.54570E-08
NU BAR = 2.43825E+00 + OR - 6.68329E-05 AVERAGE FISSION GROUP = 2.22583E+01 + OR - 3.88626E-03
ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 1.89263E-01 + OR - 6.21076E-04

NO. OF INITIAL

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.91902	+ OR - 0.00085	0.91817 TO 0.91988	0.91731 TO 0.92073	0.91646 TO 0.92158	800000
4	0.91907	+ OR - 0.00085	0.91822 TO 0.91992	0.91736 TO 0.92078	0.91651 TO 0.92163	799000
5	0.91909	+ OR - 0.00085	0.91824 TO 0.91995	0.91738 TO 0.92080	0.91653 TO 0.92165	798000
6	0.91917	+ OR - 0.00085	0.91832 TO 0.92002	0.91746 TO 0.92087	0.91661 TO 0.92173	797000
7	0.91919	+ OR - 0.00085	0.91833 TO 0.92004	0.91748 TO 0.92089	0.91663 TO 0.92175	796000
8	0.91927	+ OR - 0.00085	0.91842 TO 0.92012	0.91757 TO 0.92097	0.91672 TO 0.92182	795000
9	0.91930	+ OR - 0.00085	0.91845 TO 0.92015	0.91760 TO 0.92100	0.91675 TO 0.92186	794000
10	0.91931	+ OR - 0.00085	0.91846 TO 0.92016	0.91761 TO 0.92101	0.91675 TO 0.92186	793000
11	0.91935	+ OR - 0.00085	0.91850 TO 0.92020	0.91765 TO 0.92106	0.91680 TO 0.92191	792000
12	0.91941	+ OR - 0.00085	0.91856 TO 0.92026	0.91771 TO 0.92111	0.91686 TO 0.92196	791000
17	0.91942	+ OR - 0.00086	0.91857 TO 0.92028	0.91771 TO 0.92114	0.91686 TO 0.92199	786000
22	0.91954	+ OR - 0.00086	0.91868 TO 0.92039	0.91783 TO 0.92125	0.91697 TO 0.92210	781000
27	0.91959	+ OR - 0.00086	0.91873 TO 0.92045	0.91787 TO 0.92131	0.91701 TO 0.92217	776000
32	0.91967	+ OR - 0.00086	0.91881 TO 0.92054	0.91795 TO 0.92140	0.91709 TO 0.92226	771000
37	0.91975	+ OR - 0.00087	0.91888 TO 0.92062	0.91802 TO 0.92148	0.91715 TO 0.92235	766000
42	0.91973	+ OR - 0.00087	0.91886 TO 0.92060	0.91799 TO 0.92147	0.91712 TO 0.92234	761000
47	0.91988	+ OR - 0.00087	0.91900 TO 0.92075	0.91813 TO 0.92162	0.91726 TO 0.92249	756000
52	0.91982	+ OR - 0.00087	0.91894 TO 0.92069	0.91807 TO 0.92156	0.91720 TO 0.92243	751000
57	0.91973	+ OR - 0.00088	0.91885 TO 0.92060	0.91798 TO 0.92148	0.91710 TO 0.92235	746000
• • •						
692	0.91879	+ OR - 0.00241	0.91637 TO 0.92120	0.91396 TO 0.92362	0.91154 TO 0.92603	111000
697	0.91958	+ OR - 0.00249	0.91709 TO 0.92207	0.91459 TO 0.92457	0.91210 TO 0.92706	106000
702	0.92013	+ OR - 0.00256	0.91757 TO 0.92269	0.91501 TO 0.92525	0.91245 TO 0.92781	101000
707	0.91941	+ OR - 0.00265	0.91675 TO 0.92206	0.91410 TO 0.92471	0.91145 TO 0.92736	96000
712	0.92037	+ OR - 0.00266	0.91772 TO 0.92303	0.91506 TO 0.92569	0.91240 TO 0.92834	91000
717	0.91955	+ OR - 0.00276	0.91679 TO 0.92230	0.91403 TO 0.92506	0.91128 TO 0.92782	86000
722	0.91973	+ OR - 0.00280	0.91692 TO 0.92253	0.91412 TO 0.92533	0.91131 TO 0.92814	81000
727	0.91983	+ OR - 0.00295	0.91687 TO 0.92278	0.91392 TO 0.92574	0.91097 TO 0.92869	76000
732	0.91969	+ OR - 0.00292	0.91677 TO 0.92262	0.91385 TO 0.92554	0.91093 TO 0.92846	71000
737	0.91957	+ OR - 0.00305	0.91652 TO 0.92262	0.91346 TO 0.92568	0.91041 TO 0.92873	66000
742	0.91911	+ OR - 0.00327	0.91584 TO 0.92239	0.91257 TO 0.92566	0.90930 TO 0.92893	61000
747	0.91832	+ OR - 0.00349	0.91482 TO 0.92181	0.91133 TO 0.92531	0.90783 TO 0.92880	56000
752	0.91721	+ OR - 0.00371	0.91349 TO 0.92092	0.90978 TO 0.92464	0.90607 TO 0.92835	51000

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

757	0.91752	+ OR - 0.00396	0.91357 TO 0.92148	0.90961 TO 0.92543	0.90566 TO 0.92939	46000
762	0.91920	+ OR - 0.00432	0.91488 TO 0.92352	0.91056 TO 0.92784	0.90623 TO 0.93216	41000
767	0.91929	+ OR - 0.00446	0.91484 TO 0.92375	0.91038 TO 0.92821	0.90592 TO 0.93267	36000
772	0.91837	+ OR - 0.00492	0.91346 TO 0.92329	0.90854 TO 0.92820	0.90363 TO 0.93312	31000
777	0.91871	+ OR - 0.00500	0.91371 TO 0.92371	0.90870 TO 0.92872	0.90370 TO 0.93372	26000
782	0.91876	+ OR - 0.00587	0.91289 TO 0.92463	0.90702 TO 0.93050	0.90115 TO 0.93637	21000
787	0.91706	+ OR - 0.00606	0.91100 TO 0.92312	0.90494 TO 0.92918	0.89888 TO 0.93524	16000
792	0.92112	+ OR - 0.00667	0.91445 TO 0.92779	0.90778 TO 0.93446	0.90111 TO 0.94113	11000
797	0.91616	+ OR - 0.00797	0.90820 TO 0.92413	0.90023 TO 0.93209	0.89226 TO 0.94006	6000NAC-STC 26
DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25						

FREQUENCY FOR GENERATIONS 4 TO 803	
0.8442 TO 0.8483	*
0.8483 TO 0.8525	***
0.8525 TO 0.8566	*****
0.8566 TO 0.8608	
0.8608 TO 0.8649	*****
0.8649 TO 0.8691	***
0.8691 TO 0.8732	*****
0.8732 TO 0.8774	*****
0.8774 TO 0.8815	*****
0.8815 TO 0.8857	*****
0.8857 TO 0.8898	*****
0.8898 TO 0.8940	*****
0.8940 TO 0.8981	*****
0.8981 TO 0.9023	*****
0.9023 TO 0.9064	*****
0.9064 TO 0.9105	*****
0.9105 TO 0.9147	*****
0.9147 TO 0.9188	*****
0.9188 TO 0.9230	*****
0.9230 TO 0.9271	*****
0.9271 TO 0.9313	*****
0.9313 TO 0.9354	*****
0.9354 TO 0.9396	*****
0.9396 TO 0.9437	*****
0.9437 TO 0.9479	*****
0.9479 TO 0.9520	*****
0.9520 TO 0.9562	*****
0.9562 TO 0.9603	*****
0.9603 TO 0.9645	*****
0.9645 TO 0.9686	*****
0.9686 TO 0.9728	*****
0.9728 TO 0.9769	****
0.9769 TO 0.9811	***
0.9811 TO 0.9852	
0.9852 TO 0.9894	
0.9894 TO 0.9935	*
0.9935 TO 0.9977	*

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

FREQUENCY FOR GENERATIONS 204 TO 803

```
0.8442 TO 0.8483 *
0.8483 TO 0.8525 *
0.8525 TO 0.8566 ***
0.8566 TO 0.8608
0.8608 TO 0.8649 **
0.8649 TO 0.8691 *
0.8691 TO 0.8732 **
0.8732 TO 0.8774 *****
0.8774 TO 0.8815 *****
0.8815 TO 0.8857 *****
0.8857 TO 0.8898 *****
0.8898 TO 0.8940 *****
0.8940 TO 0.8981 *****
0.8981 TO 0.9023 *****
0.9023 TO 0.9064 *****
0.9064 TO 0.9105 *****
0.9105 TO 0.9147 *****
0.9147 TO 0.9188 *****
0.9188 TO 0.9230 *****
0.9230 TO 0.9271 *****
0.9271 TO 0.9313 *****
0.9313 TO 0.9354 *****
0.9354 TO 0.9396 *****
0.9396 TO 0.9437 *****
0.9437 TO 0.9479 *****
0.9479 TO 0.9520 *****
0.9520 TO 0.9562 *****
0.9562 TO 0.9603 *****
0.9603 TO 0.9645 ****
0.9645 TO 0.9686 ***
0.9686 TO 0.9728 *****
0.9728 TO 0.9769 ***
0.9769 TO 0.9811 ***
0.9811 TO 0.9852
0.9852 TO 0.9894
0.9894 TO 0.9935 *
0.9935 TO 0.9977 *
```

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

```

                                FREQUENCY FOR GENERATIONS 404 TO 803
0.8442 TO 0.8483      *
0.8483 TO 0.8525      *
0.8525 TO 0.8566      **
0.8566 TO 0.8608      *
0.8608 TO 0.8649      *
0.8649 TO 0.8691      *
0.8691 TO 0.8732      **
0.8732 TO 0.8774      *****
0.8774 TO 0.8815      *****
0.8815 TO 0.8857      *****
0.8857 TO 0.8898      *****
0.8898 TO 0.8940      *****
0.8940 TO 0.8981      *****
0.8981 TO 0.9023      *****
0.9023 TO 0.9064      *****
0.9064 TO 0.9105      *****
0.9105 TO 0.9147      *****
0.9147 TO 0.9188      *****
0.9188 TO 0.9230      *****
0.9230 TO 0.9271      *****
0.9271 TO 0.9313      *****
0.9313 TO 0.9354      *****
0.9354 TO 0.9396      *****
0.9396 TO 0.9437      *****
0.9437 TO 0.9479      *****
0.9479 TO 0.9520      *****
0.9520 TO 0.9562      *****
0.9562 TO 0.9603      *****
0.9603 TO 0.9645      ****
0.9645 TO 0.9686      **
0.9686 TO 0.9728      *****
0.9728 TO 0.9769      ***
0.9769 TO 0.9811      **
0.9811 TO 0.9852      *
0.9852 TO 0.9894      *
0.9894 TO 0.9935      *
0.9935 TO 0.9977      *
```

Figure 6.7-2 CSAS25 Input/Output for Directly Loaded Fuel Accident Conditions Criticality Analysis (continued)

NAC-STC 26 DIRECTLY LOADED, WEST 17X17 OFA, ACCIDENT TRANSPORT 1.0 1.0, PITCH 25

FREQUENCY FOR GENERATIONS 604 TO 803

0.8442 TO 0.8483	*
0.8483 TO 0.8525	
0.8525 TO 0.8566	**
0.8566 TO 0.8608	
0.8608 TO 0.8649	*
0.8649 TO 0.8691	*
0.8691 TO 0.8732	*
0.8732 TO 0.8774	***
0.8774 TO 0.8815	***
0.8815 TO 0.8857	****
0.8857 TO 0.8898	****
0.8898 TO 0.8940	*****
0.8940 TO 0.8981	*****
0.8981 TO 0.9023	*****
0.9023 TO 0.9064	*****
0.9064 TO 0.9105	*****
0.9105 TO 0.9147	*****
0.9147 TO 0.9188	*****
0.9188 TO 0.9230	*****
0.9230 TO 0.9271	*****
0.9271 TO 0.9313	*****
0.9313 TO 0.9354	*****
0.9354 TO 0.9396	*****
0.9396 TO 0.9437	*****
0.9437 TO 0.9479	*****
0.9479 TO 0.9520	*****
0.9520 TO 0.9562	*****
0.9562 TO 0.9603	*****
0.9603 TO 0.9645	**
0.9645 TO 0.9686	*
0.9686 TO 0.9728	
0.9728 TO 0.9769	*
0.9769 TO 0.9811	
0.9811 TO 0.9852	
0.9852 TO 0.9894	
0.9894 TO 0.9935	
0.9935 TO 0.9977	

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions

```
=CSAS25
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 cm) (EXT. MOD. VF = 0.6)
'
  THIS IS A MODEL OF THE NAC-MPC BASKET
'
  LOADED WITH 36 UNITED NUCLEAR TYPE A ASSEMBLIES
'
  INTERIOR MODERATOR VOLUME FRACTION = 0.0001
'
  EXTERIOR MODERATOR VOLUME FRACTION = 0.6
'
  CASK TO CASK PITCH = 250.698 cm
27GROUPNDF4 LATTICECELL
UO2      1      0.95      293.0 92235 4.0 92238 96.0 END
ZIRCALLOY 2      1.0      293.0      END
H2O      3      0.0001      293.0      END
AL       4      1.0      293.0      END
SS304    5      1.0      293.0      END
B-10     6  DEN=2.6226 0.0450 293.0      END
B-11     6  DEN=2.6226 0.2736 293.0      END
C        6  DEN=2.6226 0.0927 293.0      END
AL       6  DEN=2.6226 0.5737 293.0      END
PB       7      1.0      293.0      END
H        8  DEN=1.6291 0.060 293.0      END
O        8  DEN=1.6291 0.425 293.0      END
C        8  DEN=1.6291 0.277 293.0      END
N        8  DEN=1.6291 0.020 293.0      END
AL       8  DEN=1.6291 0.214 293.0      END
B-10     8  DEN=1.6291 0.001 293.0      END
B-11     8  DEN=1.6291 0.004 293.0      END
H2O      9      0.6      293.0      END
END COMP
SQUAREPITCH 1.1887 0.7887 1 3 0.9271 2 0.8052 0 END
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 cm) (EXT. MOD. VF = 0.6)
READ PARAM RUN=YES PLT=NO GEN=203 NPG=1000 TME=500 END PARAM
READ GEOM
'
'   WATER LEVEL UNIT CELLS
'
UNIT 1
COM='FUEL PIN CELL - BETWEEN DISKS'
CYLINDER 1 1 0.3943 2P2.1400
CYLINDER 0 1 0.4026 2P2.1400
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 2
COM='WATER CELL - BETWEEN DISKS'
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 3
COM='DISPLACEMENT CELL - BETWEEN DISKS'
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 4
COM='INSTRUMENT TUBE CELL - BETWEEN DISKS'
CYLINDER 3 1 0.4998 2P2.1400

CYLINDER 5 1 0.5442 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
'
'   DISK LEVEL UNIT CELLS (BOTH SS AND AL)
'
UNIT 5
COM='FUEL PIN CELL - WITH SS DISK'
CYLINDER 1 1 0.3943 2P0.6604
CYLINDER 0 1 0.4026 2P0.6604
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 6
COM='WATER CELL - WITH SS DISK'
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 7
COM='DISPLACEMENT CELL - WITH SS DISK'
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 8
COM='INSTRUMENT TUBE CELL - WITH SS DISK'
CYLINDER 3 1 0.4998 2P0.6604
CYLINDER 5 1 0.5442 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
'
'   WATER LEVEL BORAL SHEETS
UNIT 14
COM='X-X BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P9.144 2P0.0318 2P2.1400
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
CUBOID 4 1 2P9.144 2P0.0953 2P2.1400
UNIT 15
COM='Y-Y BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P0.0318 2P9.144 2P2.1400
CUBOID 4 1 2P0.0953 2P9.144 2P2.1400
' DISK LEVEL BORAL SHEETS (AL AND SS)
UNIT 16
COM='X-X BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P9.144 2P0.0318 2P0.6604
CUBOID 4 1 2P9.144 2P0.0953 2P0.6604
UNIT 17
COM='Y-Y BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P0.0318 2P9.144 2P0.6604
CUBOID 4 1 2P0.0953 2P9.144 2P0.6604
' WATER LEVEL WEB MATERIAL
UNIT 20
COM='WATER LEVEL WEB MATERIAL (SMALL) X-X'
CUBOID 3 1 2P10.4635 2P0.9716 2P2.1400
UNIT 21
COM='WATER LEVEL WEB MATERIAL (MEDIUM) X-X'
CUBOID 3 1 2P10.4635 2P1.0478 2P2.1400
UNIT 22
COM='WATER LEVEL WEB MATERIAL (LARGE) X-X'
CUBOID 3 1 2P10.4635 2P1.1208 2P2.1400
UNIT 23
COM='WATER LEVEL WEB MATERIAL (LONG) Y-Y'
CUBOID 3 1 2P1.1208 2P79.5249 2P2.1400
' SUPPORT DISK WEB MATERIAL
'
UNIT 30
COM='SUPPORT DISK WEB MATERIAL (SMALL) X-X'
CUBOID 5 1 2P10.4635 2P0.9716 2P0.6604
UNIT 31
COM='SUPPORT DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 5 1 2P10.4635 2P1.0478 2P0.6604
UNIT 32
COM='SUPPORT DISK WEB MATERIAL (LARGE) X-X'
CUBOID 5 1 2P10.4635 2P1.1208 2P0.6604
UNIT 33
COM='SUPPORT DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 5 1 2P1.1208 2P79.5249 2P0.6604
'
' HEAT TRANSFER DISK WEB MATERIAL
'
UNIT 40
COM='HEAT TRANSFER DISK WEB MATERIAL (SMALL) X-X'
CUBOID 4 1 2P10.4635 2P0.9716 2P0.6604
UNIT 41
COM='HEAT TRANSFER DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 4 1 2P10.4635 2P1.0478 2P0.6604
UNIT 42
COM='HEAT TRANSFER DISK WEB MATERIAL (LARGE) X-X'
CUBOID 4 1 2P10.4635 2P1.1208 2P0.6604
UNIT 43
COM='HEAT TRANSFER DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 4 1 2P1.1208 2P79.5249 2P0.6604
' WATER LEVEL ASSEMBLY ARRAYS
UNIT 50
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL'
ARRAY 1 -9.5104 -9.5104 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400
UNIT 51
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.0 -0.1584 0.0
UNIT 52
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
```


Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```

CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 0.0 0.1584 0.0
UNIT 53
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 -0.1584 0.0 0.0
UNIT 54
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 0.1584 0.0 0.0
UNIT 55
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 0.1584 0.1584 0.0
UNIT 56
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 -0.1584 0.1584 0.0
UNIT 57
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 0.1584 -0.1584 0.0
UNIT 58
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 50 -0.1584 -0.1584 0.0
UNIT 59
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636      2P2.1400
'
SUPPORT DISK LEVEL ASSEMBLY ARRAYS
'
UNIT 60
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441      2P0.6604
CUBOID 5 1 4P10.0661      2P0.6604
CUBOID 3 1 4P10.25681      2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051      2P0.6604
UNIT 61
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 0.0 -0.1584 0.0
UNIT 62
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 0.0 0.1584 0.0
UNIT 63
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 -0.1584 0.0 0.0
UNIT 64
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 0.1584 0.0 0.0
UNIT 65
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 0.1584 0.1584 0.0
UNIT 66
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 -0.1584 0.1584 0.0
UNIT 67
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 0.1584 -0.1584 0.0
UNIT 68
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 60 -0.1584 -0.1584 0.0
UNIT 69
COM='CENTRAL HOLE'

```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
CUBOID 3 1 4P10.4636 2P0.6604
'
' HEAT TRANSFER DISK LEVEL ASSEMBLY ARRAYS
'
UNIT 70
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
UNIT 71
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.0 -0.1584 0.0
UNIT 72
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.0 0.1584 0.0
UNIT 73
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 -0.1584 0.0 0.0
UNIT 74
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.1584 0.0 0.0
UNIT 75
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.1584 0.1584 0.0
UNIT 76
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 -0.1584 0.1584 0.0
UNIT 77
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.1584 -0.1584 0.0
UNIT 78
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 -0.1584 -0.1584 0.0
UNIT 79
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P0.6604
'
' WATER LEVEL BASKET ARRAYS
'
UNIT 80
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 20 -10.4636 -33.6323 -2.1400
UNIT 81
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 21 -10.4636 -33.6323 -2.1400
UNIT 82
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 22 -10.4636 -56.6549 -2.1400
UNIT 83
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 23 -10.4636 -56.6549 -2.1400
UNIT 84
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 24 -10.4636 -79.5251 -2.1400
UNIT 85
COM='13X1 WATER LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 25 -10.4636 -79.5251 -2.1400
UNIT 86
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 26 -10.4636 -79.5251 -2.1400
'
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
' SUPPORT DISK LEVEL BASKET ARRAYS
'
UNIT 90
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 30 -10.4636 -33.6323 -0.6604
UNIT 91
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 31 -10.4636 -33.6323 -0.6604
UNIT 92
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 32 -10.4636 -56.6549 -0.6604
UNIT 93
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 33 -10.4636 -56.6549 -0.6604
UNIT 94
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 34 -10.4636 -79.5251 -0.6604
UNIT 95
COM='13X1 SUPPORT DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 35 -10.4636 -79.5251 -0.6604
UNIT 96
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 36 -10.4636 -79.5251 -0.6604
'
' HEAT TRANSFER DISK LEVEL BASKET ARRAYS
'
UNIT 100
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 40 -10.4636 -33.6323 -0.6604
UNIT 101
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 41 -10.4636 -33.6323 -0.6604
UNIT 102
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 42 -10.4636 -56.6549 -0.6604
UNIT 103
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 43 -10.4636 -56.6549 -0.6604
UNIT 104
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 44 -10.4636 -79.5251 -0.6604
UNIT 105
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 45 -10.4636 -79.5251 -0.6604
UNIT 106
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 46 -10.4636 -79.5251 -0.6604
'
' BASKET ARRAY IN TRANSPORT CASK OVERPACK (LEVEL CONSTRUCTION)
'
UNIT 110
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - WATER LEVEL'
ARRAY 50 -33.6323 -79.5251 -2.1400
CYLINDER 3 1 88.1253 2P2.1400
HOLE 80 -69.0614 0.0 0.0
HOLE 82 -46.1912 0.0 0.0
HOLE 81 69.0614 0.0 0.0
HOLE 83 46.1912 0.0 0.0
CYLINDER 5 1 89.7128 2P2.1400
CYLINDER 3 1 90.170 2P2.1400
CYLINDER 5 1 93.98 2P2.1400
CYLINDER 7 1 103.4288 2P2.1400
CYLINDER 5 1 110.109 2P2.1400
CYLINDER 8 1 124.714 2P2.1400
CYLINDER 5 1 125.349 2P2.1400
CUBOID 9 1 4P125.349 2P2.1400
UNIT 111
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - SUPPORT DISK LEVEL'
ARRAY 51 -33.6323 -79.5251 -0.6604
CYLINDER 5 1 87.6046 2P0.6604
HOLE 90 -69.0614 0.0 0.0
HOLE 92 -46.1912 0.0 0.0
HOLE 91 69.0614 0.0 0.0
HOLE 93 46.1912 0.0 0.0
CYLINDER 3 1 88.1253 2P0.6604
CYLINDER 5 1 89.7128 2P0.6604
CYLINDER 3 1 90.170 2P0.6604
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
CYLINDER 5 1 93.98      2P0.6604
CYLINDER 7 1 103.4288   2P0.6604
CYLINDER 5 1 110.109    2P0.6604
CYLINDER 8 1 124.714    2P0.6604
CYLINDER 5 1 125.349    2P0.6604
CUBOID 9 1 4P125.349    2P0.6604
UNIT 112
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - HEAT TRANSFER DISK LEVEL'
ARRAY 52 -33.6323 -79.5251 -0.6604
CYLINDER 4 1 87.2490    2P0.6604
HOLE 100 -69.0614 0.0 0.0
HOLE 102 -46.1912 0.0 0.0
HOLE 101 69.0614 0.0 0.0
HOLE 103 46.1912 0.0 0.0
CYLINDER 3 1 88.1253    2P0.6604
CYLINDER 5 1 89.7128    2P0.6604
CYLINDER 3 1 90.170     2P0.6604
CYLINDER 5 1 93.98      2P0.6604
CYLINDER 7 1 103.4288   2P0.6604
CYLINDER 5 1 110.109    2P0.6604
CYLINDER 8 1 124.714    2P0.6604
CYLINDER 5 1 125.349    2P0.6604
CUBOID 9 1 4P125.349    2P0.6604

1
1 GLOBAL UNIT
1
GLOBAL UNIT 120
ARRAY 60 -175.349 -175.349 0.0
END GEOM
READ ARRAY
ARA=1 NUX=16 NUY=16 NUZ=1 FILL
1 1 1 1 1 1 1 1 3 2 2 2 2 2 2 2 2
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
3 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 4 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
END FILL
ARA=2 NUX=16 NUY=16 NUZ=1 FILL
5 5 5 5 5 5 5 5 7 6 6 6 6 6 6 6 6
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
7 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 8 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
END FILL
1
1 WATER LEVEL ARRAYS
1
ARA=20 NUX=1 NUY=5 NUZ=1
FILL
55
22
54
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
22
57
END FILL
ARA=21 NUX=1 NUY=5 NUZ=1
FILL
56
22
53
22
58
END FILL
ARA=22 NUX=1 NUY=9 NUZ=1
FILL
55
21
55
22
54
22
57
21
57
END FILL
ARA=23 NUX=1 NUY=9 NUZ=1
FILL
21
56
22
53
22
58
21
58
END FILL
ARA=24 NUX=1 NUY=13 NUZ=1
FILL
55
20
55
21
55
22
54
22
57
21
57
20
57
END FILL
ARA=25 NUX=1 NUY=13 NUZ=1
FILL
52
20
52
21
52
22
59
22
51
21
51
20
51
END FILL
ARA=26 NUX=1 NUY=13 NUZ=1
FILL
56
20
56
21
56
22
53
22
58
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
21
58
20
58
END FILL
'
' SUPPOR DISK LEVEL ARRAYS
'
ARA=30 NUX=1 NUY=5 NUZ=1
FILL
65
32
64
32
67
END FILL
ARA=31 NUX=1 NUY=5 NUZ=1
FILL
66
32
63
32
68
END FILL
ARA=32 NUX=1 NUY=9 NUZ=1
FILL
65
31
65
32
64
32
67
31
67
END FILL
ARA=33 NUX=1 NUY=9 NUZ=1
FILL
66
31
66
32
63
32
68
31
68
END FILL
ARA=34 NUX=1 NUY=13 NUZ=1
FILL
65
30
65
31
65
32
64
32
67
31
67
30
67
END FILL
ARA=35 NUX=1 NUY=13 NUZ=1
FILL
62
30
62
31
62
32
69
32
61
31
61
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
30
61
END FILL
ARA=36 NUX=1 NUY=13 NUZ=1
FILL
66
30
66
31
66
32
63
32
68
31
68
30
68
END FILL
HEAT TRANSFER DISK LEVEL ARRAYS
ARA=40 NUX=1 NUY=5 NUZ=1
FILL
75
42
74
42
77
END FILL
ARA=41 NUX=1 NUY=5 NUZ=1
FILL
76
42
73
42
78
END FILL
ARA=42 NUX=1 NUY=9 NUZ=1
FILL
75
41
75
42
74
42
77
41
77
END FILL
ARA=43 NUX=1 NUY=9 NUZ=1
FILL
76
41
76
42
73
42
78
41
78
END FILL
ARA=44 NUX=1 NUY=13 NUZ=1
FILL
75
40
75
41
75
42
74
42
77
41
77
40
77
```

Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
END FILL
ARA=45 NUX=1 NUY=13 NUZ=1
FILL
72
40
72
41
72
42
79
42
71
41
71

40
71
END FILL
ARA=46 NUX=1 NUY=13 NUZ=1
FILL
76
40
76
41
76
42
73
42
78
41
78
40
78
END FILL
' MAJOR ARRAYS
'
ARA=50 NUX=5 NUY=1 NUZ=1
FILL
84 23 85 23 86
END FILL
ARA=51 NUX=5 NUY=1 NUZ=1
FILL
94 33 95 33 96
END FILL
ARA=52 NUX=5 NUY=1 NUZ=1
FILL
104 43 105 43 106
END FILL
' GLOBAL ARRAY
'
ARA=60 NUX=1 NUY=1 NUZ=4
FILL
112
110
111
110
END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIR END BOUNDS
READ PLOT
SCR=YES PIC=MAT LPI=10
UAX=1.0 VDN=-1.0 NAX=1500
'
' WHOLE BASKET HORIZONTAL SLICES
'
TTL='BASKET X-Y CROSS SECTION AT Z= 0.635 HEAT TRANSFER DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 0.635
XLR= 130 YLR= -130 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 3.44 WATER LEVEL'
XUL= -130 YUL= 130 ZUL= 3.44
XLR= 130 YLR= -130 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 6.236 SS DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 6.236
```


Figure 6.7-3 CSAS25 Input for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
XLR= 130 YLR= -130 ZLR= 6.236
UAX=1.0 VDN=-1.0 NAX=1500 END

' HEAT TRANSFER DISK LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I HEAT TRANSFER DISK'
XUL= 12. YUL= 80 ZUL= 0.635
XLR= 80.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II HEAT TRANSFER DISK'
XUL= 12.0 YUL= -12.0 ZUL= 0.635
XLR= 80 YLR= -80 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III HEAT TRANSFER DISK'
XUL= -80.0 YUL= -12.0 ZUL= 0.635
XLR= -12.0 YLR= -80.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV HEAT TRANSFER DISK'
XUL= -80.0 YUL= 80.0 ZUL= 0.635
XLR= -12.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END

' WATER LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I WATER LEVEL'
XUL= 12. YUL= 80 ZUL= 3.44
XLR= 80.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II WATER LEVEL'
XUL= 12.0 YUL= -12.0 ZUL= 3.44
XLR= 80 YLR= -80 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III WATER LEVEL'
XUL= -80.0 YUL= -12.0 ZUL= 3.44
XLR= -12.0 YLR= -80.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV WATER LEVEL'
XUL= -80.0 YUL= 80.0 ZUL= 3.44
XLR= -12.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END

' VERTICAL SLICES

TTL='BASKET X-Z CROSS SECTION ALUMINUM LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 1.27
XLR= 90 YLR=0.4 ZLR= -.1
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION WATER LEVEL (MIDDLE OF FUEL PIN)'

XUL= -90 YUL=0.4 ZUL= 4.318
XLR= 90 YLR=0.4 ZLR= 1.27
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION SS LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 6.858
XLR= 90 YLR=0.4 ZLR= 5.588
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION ENTIRE MODEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 12
XLR= 90 YLR=0.4 ZLR= 0
UAX=1.0 WDN=-1.0 NAX=1500 END
END PLOT
END DATA
END
```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions

```
*****
*****
*****      PROGRAM VERIFICATION INFORMATION      *****
*****
*****      CODE SYSTEM:  SCALE-PC VERSION:  4.3      *****
*****
*****
*****
*****      PROGRAM:  CSAS      *****
*****
*****      CREATION DATE:  03-08-96      *****
*****
*****      VOLUME:  ENG      *****
*****
*****      LIBRARY:  G:\scale43\exe      *****
*****
*****      PRODUCTION CODE:  CSAS      *****
*****
*****      VERSION:  3.1      *****
*****
*****      JOBNAME:  SCALE-PC      *****
*****
*****      DATE OF EXECUTION:  11/13/96      *****
*****
*****      TIME OF EXECUTION:  00:51:47      *****
*****
*****
*****
*****
```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.

**** PROBLEM PARAMETERS ****

LIB 27GROUPNDF4 LIBRARY
MXK 9 MIXTURES
MSC 18 COMPOSITION SPECIFICATIONS
IZM 4 MATERIAL ZONES
GE LATTICECELL GEOMETRY
MORE 0 0/1 DO NOT READ/READ OPTIONAL PARAMETER DATA
MSLN 0 FUEL SOLUTIONS

**** PROBLEM COMPOSITION DESCRIPTION ****

SC UO2 STANDARD COMPOSITION
MX 1 MIXTURE NO.
VF 0.9500 VOLUME FRACTION
ROTH 10.9600 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
92000 1.00 ATOM/MOLECULE
92235 4.000 WT%
92238 96.000 WT%
8016 2.00 ATOMS/MOLECULE
END

SC ZIRCALLOY STANDARD COMPOSITION
MX 2 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 6.5600 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
40302 1.00 ATOM/MOLECULE
END

SC H2O STANDARD COMPOSITION
MX 3 MIXTURE NO.
VF 0.0001 VOLUME FRACTION
ROTH 0.9982 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
1001 2.00 ATOMS/MOLECULE
8016 1.00 ATOM/MOLECULE
END

SC AL STANDARD COMPOSITION
MX 4 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 2.7020 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
ICP      1 0/1 MIXTURE/COMPOUND
TEMP     293.0 DEG KELVIN
        13027      1.00 ATOM/MOLECULE
END
```

```
SC SS304      STANDARD COMPOSITION
MX           5 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        7.9200 THEORETICAL DENSITY
NEL          4 NO. ELEMENTS
ICP          0 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
        24304      19.000 WT%
        25055       2.000 WT%
        26304      69.500 WT%
        28304       9.500 WT%
END
```

```
SC B-10      STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.0450 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
        5010       1.00 ATOM/MOLECULE
END
```

```
SC B-11      STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.2736 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
        5011       1.00 ATOM/MOLECULE
END
```

```
SC C         STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.0927 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
        6012       1.00 ATOM/MOLECULE
END
```

```
SC AL        STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.5737 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

13027 1.00 ATOM/MOLECULE
END

SC PB STANDARD COMPOSITION
MX 7 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 11.3440 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
82000 1.00 ATOM/MOLECULE
END

SC H STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.0600 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
1001 1.00 ATOM/MOLECULE
END

SC O STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.4250 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
8016 1.00 ATOM/MOLECULE
END

SC C STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.2770 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
6012 1.00 ATOM/MOLECULE
END

SC N STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.0200 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
7014 1.00 ATOM/MOLECULE
END

SC AL STANDARD COMPOSITION
MX 8 MIXTURE NO.

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
VF      0.2140 VOLUME FRACTION
ROTH    1.6291 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        13027      1.00 ATOM/MOLECULE
END
```

```
SC B-10      STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF      0.0010 VOLUME FRACTION
ROTH    1.6291 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5010      1.00 ATOM/MOLECULE
END
```

```
SC B-11      STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF      0.0040 VOLUME FRACTION
ROTH    1.6291 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5011      1.00 ATOM/MOLECULE
END
```

```
SC H2O       STANDARD COMPOSITION
MX           9 MIXTURE NO.
VF      0.6000 VOLUME FRACTION
ROTH    0.9982 THEORETICAL DENSITY
NEL      2 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        1001      2.00 ATOMS/MOLECULE
        8016      1.00 ATOM/MOLECULE
END
```

**** PROBLEM GEOMETRY ****

```
CTP SQUAREPITCH CELL TYPE
PITCH    1.1887 CM CENTER TO CENTER SPACING
FUELOD   0.7887 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL     1 MIXTURE NO. OF FUEL
MMOD      3 MIXTURE NO. OF MODERATOR
CLADOD   0.9271 CM CLAD OUTER DIAMETER
MCLAD     2 MIXTURE NO. OF CLAD
GAPOD    0.8052 CM GAP OUTER DIAMETER
MGAP      0 MIXTURE NO. OF GAP
```

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD

```
*****
*****
*****      PROGRAM VERIFICATION INFORMATION      *****
*****
*****      CODE SYSTEM:  SCALE-PC VERSION:  4.3      *****
*****
*****
*****
*****      PROGRAM:  000002      *****
*****
*****      CREATION DATE:  09-28-95      *****
*****
*****      VOLUME:  ENG      *****
*****
*****      LIBRARY:  G:\scale43\exe      *****
*****
*****
*****      PRODUCTION CODE:  NITAWL      *****
*****
*****      VERSION:  3.0      *****
*****
*****      JOBNAME:  SCALE-PC      *****
*****
*****      DATE OF EXECUTION:  11/13/96      *****
*****
*****      TIME OF EXECUTION:  00:51:51      *****
*****
*****
*****
*****
```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
-1Q ARRAY HAS      1 ENTRIES.

0Q ARRAY HAS      9 ENTRIES.

1Q ARRAY HAS     12 ENTRIES.

SELECT 25 NUCLIDES FROM THE MASTER LIBRARY ON LOGICAL 1
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL 2
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL 3
      TO CREATE THE NEW WORKING LIBRARY ON LOGICAL 4

      4 RESONANCE CALCULATIONS HAVE BEEN REQUESTED
      -1 OUTPUT OPTION FOR AMPX FORMATTED CROSS SECTION DATA
      2001 MAXIMUM NUMBER OF RESONANCE MESH INTERVALS
      2 ORDER OF RESONANCE LEVEL PROCESSING

THE STORAGE ALLOCATED FOR THIS CASE IS 100000 WORDS

2Q ARRAY HAS     25 ENTRIES.

3Q ARRAY HAS     60 ENTRIES.

4Q ARRAY HAS     25 ENTRIES.

GENERAL INFORMATION CONCERNING CROSS SECTION LIBRARY
TAPE IDENTIFICATION NUMBER      4321
NUMBER OF NUCLIDES ON TAPE      26
NUMBER OF NEUTRON ENERGY GROUPS 27
FIRST THERMAL NEUTRON ENERGY GROUP 15
NUMBER OF GAMMA ENERGY GROUPS  0

DIRECT ACCESS UNIT NUMBER 9 REQUIRES 117 BLOCKS OF LENGTH 1680 WORDS
XSDRN TAPE 4321

SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
BASED ON ENDF-B VERSION 4 DATA
COMPILED FOR NRC 1/27/89
LAST UPDATED
L.M.PETRIE - ORNL

08/12/94

NUCLIDES FROM XSDRN TAPE
1 HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 3001001
2 HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 8001001
3 HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED 08/12/94 9001001
4 B-10 1273 218NGP 042375 P-3 293K UPDATED 08/12/94 6005010
5 B-10 1273 218NGP 042375 P-3 293K UPDATED 08/12/94 8005010
6 BORON-11 ENDF/B-IV MAT 1160 UPDATED 08/12/94 6005011
7 BORON-11 ENDF/B-IV MAT 1160 UPDATED 08/12/94 8005011
8 CARBON-12 ENDF/B-IV MAT 1274/THRM1065 UPDATED 08/12/94 6006012
9 CARBON-12 ENDF/B-IV MAT 1274/THRM1065 UPDATED 08/12/94 8006012
10 NITROGEN-14 ENDF/B-IV MAT 1275 UPDATED 08/12/94 8007014
11 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 1008016
12 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 3008016
13 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 8008016
14 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 9008016
15 AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 4013027
```


Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

16	AL-27 1193 218 GP 040375(5)	UPDATED 08/12/94	6013027	
17	AL-27 1193 218 GP 040375(5)	UPDATED 08/12/94	8013027	
18	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94	5024304	
19	MANGANESE-55 ENDF/B-IV MAT 1197	UPDATED 08/12/94	5025055	
20	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94	5026304	
21	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94	5028304	
22	ZIRCALLOY ENDF/B-IV MAT 1284	UPDATED 08/12/94	2040302	
23	PB 1288 218NGP 042375 P-3 293K	UPDATED 08/12/94	7082000	
24	URANIUM-235 ENDF/B-IV MAT 1261	UPDATED 08/12/94	1092235	
25	URANIUM-238 ENDF/B-IV MAT 1262	UPDATED 08/12/94	1092238	
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	3008016	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	8008016	TEMPERATURE= 293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED 08/12/94 9008016 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 4013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 6013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5) UPDATED 08/12/94 8013027 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5024304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

MANGANESE-55 ENDF/B-IV MAT 1197 UPDATED 08/12/94 5025055 TEMPERATURE= 293.00

GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	54.466	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	2.590	LUMPED NUCLEAR DENSITY	=	1.7363295E-03
SPIN FACTOR (G)	=	14.448	LUMP DIMENSION (A-BAR)	=	0.0000000E+00
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	0.0000000E+00

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 55.845 SIGMA (PER ABSORBER ATOM) = 3.4663022E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 55.925 SIGMA (PER ABSORBER ATOM) = 1.2557598E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-5.518788E-04	0.000000E+00	-3.944190E-01
9	-2.797993E-03	0.000000E+00	-2.293471E+00
10	-3.291452E-01	0.000000E+00	-3.820862E+01
11	-2.680562E+00	0.000000E+00	-1.159996E+02

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 3.33719E+00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

FISSION 0.00000E+00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 5026304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 5028304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
ZIRCALLOY ENDF/B-IV MAT 1284 UPDATED 08/12/94 2040302 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 90.436 TEMPERATURE(KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 6.385 LUMPED NUCLEAR DENSITY = 4.3307818E-02
SPIN FACTOR (G) = 1.079 LUMP DIMENSION (A-BAR) = 4.6355000E-01
INNER RADIUS = 4.0259999E-01 DANCORFF CORRECTION (C) = 8.4757560E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-8.795965E-04	0.000000E+00	-6.970308E-01
9	-4.269687E-02	0.000000E+00	-1.811447E+00
10	-4.621373E-02	0.000000E+00	-1.032856E+00
11	-1.360111E-01	0.000000E+00	-6.097271E-01

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 3.93040E-01
FISSION 0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
PB 1288 218NGP 042375 P-3 293K UPDATED 08/12/94 7082000 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
URANIUM-235 ENDF/B-IV MAT 1261 UPDATED 08/12/94 1092235 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 233.025 TEMPERATURE(KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 11.500 LUMPED NUCLEAR DENSITY = 9.4064139E-04
SPIN FACTOR (G) = 15171.100 LUMP DIMENSION (A-BAR) = 3.9434999E-01

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

INNER RADIUS = 0.0000000E+00 DANC OFF CORRECTION (C) = 9.5888901E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA(PER ABSORBER ATOM) = 1.9199110E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 238.051 SIGMA(PER ABSORBER ATOM) = 2.9209552E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
12	-1.115460E+01	-6.910128E+00	-1.682605E-01
13	-2.879944E+01	-1.404006E+01	-4.031188E-01
14	-2.298467E+01	-1.317284E+01	-4.264960E-02
15	-3.215079E-03	-2.431171E-03	3.882847E-05

EXCESS RESONANCE INTEGRALS

	RESOLVED
ABSORPTION	1.49929E+02
FISSION	9.24167E+01

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-238 ENDF/B-IV MAT 1262 UPDATED 08/12/94 1092238 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 236.006 TEMPERATURE (KELVIN) = 293.000

POTENTIAL SCATTER SIGMA = 10.599 LUMPED NUCLEAR DENSITY = 2.2290209E-02

SPIN FACTOR (G) = 656.527 LUMP DIMENSION (A-BAR) = 3.9434999E-01

INNER RADIUS = 0.0000000E+00 DANC OFF CORRECTION (C) = 9.5888901E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA(PER ABSORBER ATOM) = 8.1019773E+00

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 235.044 SIGMA(PER ABSORBER ATOM) = 5.0228214E-01

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

GROUP	RES ABS	RES FISS	RES SCAT
9	-8.035530E-02	0.000000E+00	-7.473816E-01
10	-1.506941E+00	-6.750685E-05	-8.777449E+00
11	-1.070745E+01	0.000000E+00	-2.894283E+01
12	-4.478590E+01	0.000000E+00	-5.118077E+01
13	-5.544063E+01	0.000000E+00	-1.758312E+01
14	-1.076790E+02	0.000000E+00	-5.655300E+00
15	-6.314348E-05	0.000000E+00	1.227886E-04

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 7.05643E+00
FISSION 4.17667E-04

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

THIS XSDRN WORKING TAPE WAS CREATED 11/13/96 AT 00:51:51

THE TITLE OF THE PARENT CASE IS AS FOLLOWS

SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY

BASED ON ENDF-B VERSION 4 DATA

COMPILED FOR NRC 1/27/89

TAPE ID	4321	NUMBER OF NUCLIDES	25
NUMBER OF NEUTRON GROUPS	27	NUMBER OF GAMMA GROUPS	0
FIRST THERMAL GROUP	15	LOGICAL UNIT	4

TABLE OF CONTENTS

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 3001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 8001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 9001001
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 6005010
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 8005010
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	ID 6005011
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	ID 8005011
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	ID 6006012
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	ID 8006012
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	ID 8007014
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 1008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 3008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 8008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 9008016
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 4013027
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 6013027
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 8013027
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5024304
MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94	ID 5025055
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5026304
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5028304
ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94	ID 2040302
PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 7082000
URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94	ID 1092235
URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94	ID 1092238

TAPE COPY USED 0 I/O'S, AND TOOK 0.33 SECONDS

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```
*****  
*****  
*****      PROGRAM VERIFICATION INFORMATION      *****  
*****  
*****      CODE SYSTEM:   SCALE-PC VERSION:   4.3      *****  
*****  
*****  
*****  
*****  
*****  
*****      PROGRAM:   000009      *****  
*****  
*****      CREATION DATE:   03-08-96      *****  
*****  
*****      VOLUME:       ENG      *****  
*****  
*****      LIBRARY:     G:\scale43\exe      *****  
*****  
*****  
*****      PRODUCTION CODE: KENOVA      *****  
*****  
*****      VERSION:    3.1      *****  
*****  
*****      JOBNAME:    SCALE-PC      *****  
*****  
*****      DATE OF EXECUTION: 11/13/96      *****  
*****  
*****      TIME OF EXECUTION: 00:52:03      *****  
*****  
*****  
*****
```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```

*****
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.)
*****
*****      NUMERIC PARAMETERS      *****
*****
TME      MAXIMUM PROBLEM TIME (MIN)      500.00
TBA      TIME PER GENERATION (MIN)      0.50
GEN      NUMBER OF GENERATIONS      203
NPG      NUMBER PER GENERATION      1000
NSK      NUMBER OF GENERATIONS TO BE SKIPPED      3
BEG      BEGINNING GENERATION NUMBER      1
RES      GENERATIONS BETWEEN CHECKPOINTS      0
X1D      NUMBER OF EXTRA 1-D CROSS SECTIONS      1
NBK      NEUTRON BANK SIZE      1025
XNB      EXTRA POSITIONS IN NEUTRON BANK      0
NFB      FISSION BANK SIZE      1000
XFB      EXTRA POSITIONS IN FISSION BANK      0
WTA      DEFAULT VALUE OF WEIGHT AVERAGE      0.5000
WTH      WEIGHT HIGH FOR SPLITTING      3.0000
WTL      WEIGHT LOW FOR RUSSIAN ROULETTE      0.3333
RND      STARTING RANDOM NUMBER      BB827100001
NB8      NUMBER OF D.A. BLOCKS ON UNIT 8      200
NL8      LENGTH OF D.A. BLOCKS ON UNIT 8      512
ADJ      MODE OF CALCULATION      FORWARD
          INPUT DATA WRITTEN ON RESTART UNIT      NO
          BINARY DATA INTERFACE      YES

```

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

```

*****
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.
***
***** LOGICAL PARAMETERS *****
***
*** RUN  EXECUTE PROBLEM AFTER CHECKING DATA  YES      PLT  PLOT PICTURE MAP(S)          NO ***
***
*** FLX  COMPUTE FLUX                          NO      FDN  COMPUTE FISSION DENSITIES      NO ***
***
*** SMU  COMPUTE AVG UNIT SELF-MULTIPLICATION  NO      NUB  COMPUTE NU-BAR & AVG FISSION GROUP  YES ***
***
*** MKU  COMPUTE MATRIX K-EFF BY UNIT NUMBER   NO      MKP  COMPUTE MATRIX K-EFF BY UNIT LOCATION NO ***
***
*** CKU  COMPUTE COFACTOR K-EFF BY UNIT NUMBER NO      CKP  COMPUTE COFACTOR K-EFF BY UNIT LOCATION NO ***
***
*** FMU  PRINT FISSION PROD MATRIX BY UNIT NUMBER NO    FMP  PRINT FISSION PROD MATRIX BY UNIT LOCATION NO ***
***
*** MKH  COMPUTE MATRIX K-EFF BY HOLE NUMBER   NO      MKA  COMPUTE MATRIX K-EFF BY ARRAY NUMBER  NO ***
***
*** CKH  COMPUTE COFACTOR K-EFF BY HOLE NUMBER NO      CKA  COMPUTE COFACTOR K-EFF BY ARRAY NUMBER  NO ***
***
*** FMH  PRINT FISSION PROD MATRIX BY HOLE NUMBER NO    FMA  PRINT FISSION PROD MATRIX BY ARRAY NUMBER NO ***
***
*** HHL  COLLECT MATRIX BY HIGHEST HOLE LEVEL  NO      HAL  COLLECT MATRIX BY HIGHEST ARRAY LEVEL  NO ***
***
*** AMX  PRINT ALL MIXED CROSS SECTIONS        NO      FAR  PRINT FIS. AND ABS. BY REGION          NO ***
***
*** XS1  PRINT 1-D MIXTURE X-SECTIONS          NO      GAS  PRINT FAR BY GROUP                    NO ***
***
*** XS2  PRINT 2-D MIXTURE X-SECTIONS          NO      PAX  PRINT XSEC-ALBEDO CORRELATION TABLES NO ***
***
*** XAP  PRINT MIXTURE ANGLES & PROBABILITIES  NO      PWT  PRINT WEIGHT AVERAGE ARRAY          NO ***
***
*** PKI  PRINT FISSION SPECTRUM                NO      PGM  PRINT INPUT GEOMETRY                  NO ***
***
*** P1D  PRINT EXTRA 1-D CROSS SECTIONS        NO      BUG  PRINT DEBUG INFORMATION              NO ***
***
***                                     TRK  PRINT TRACKING INFORMATION          NO ***
***
*****

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Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

PARAMETER INPUT COMPLETED

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD = 3.0E-05

MIXTURE = 1										DENSITY(G/CC) = 10.412	
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE						
1008016	4.64617E-02	1.18487E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276			UPDATED		
08/12/94											
1092235	9.40641E-04	3.52606E-02	92235	235.0441	URANIUM-235	ENDF/B-IV MAT 1261			UPDATED		
08/12/94											
1092238	2.22902E-02	8.46253E-01	92238	238.0510	URANIUM-238	ENDF/B-IV MAT 1262			UPDATED		
08/12/94											
MIXTURE = 2										DENSITY(G/CC) = 6.5600	
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE						
2040302	4.33078E-02	1.00000E+00	40000	91.2196	ZIRCALLOY	ENDF/B-IV MAT 1284			UPDATED		
08/12/94											
MIXTURE = 3										DENSITY(G/CC) = 0.99817E-04	
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE						
3001001	6.67692E-06	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002			UPDATED		
08/12/94											
3008016	3.33846E-06	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276			UPDATED		
08/12/94											
MIXTURE = 4										DENSITY(G/CC) = 2.7020	
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE						
4013027	6.03066E-02	1.00000E+00	13027	26.9818	AL-27 1193 218 GP 040375(5)				UPDATED		
08/12/94											
MIXTURE = 5										DENSITY(G/CC) = 7.9200	
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE						

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

5024304	1.74286E-02	1.90000E-01	24000	51.9957	CR 1191 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)	UPDATED
08/12/94						
5025055	1.73633E-03	1.99999E-02	25055	54.9379	MANGANESE-55 ENDF/B-IV MAT 1197	UPDATED
08/12/94						
5026304	5.93579E-02	6.95000E-01	26000	55.8447	FE 1192 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)	UPDATED
08/12/94						
5028304	7.72070E-03	9.50001E-02	28000	58.6872	NI 1190 WT SS-304 (1/EST) P-3 293K SP=5+4 (42375)	UPDATED
08/12/94						
MIXTURE = 6 DENSITY(G/CC) = 2.5833						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
6005010	7.09799E-03	4.56855E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
6005011	3.92499E-02	2.77771E-01	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
6006012	1.22006E-02	9.41116E-02	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
6013027	3.35812E-02	5.82432E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE = 7 DENSITY(G/CC) = 11.344						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K	UPDATED
08/12/94						
MIXTURE = 8 DENSITY(G/CC) = 1.6307						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
8001001	5.84084E-02	5.99323E-02	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
8005010	9.79802E-05	9.99025E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
8005011	3.56450E-04	3.99615E-03	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
8006012	2.26463E-02	2.76729E-01	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
8007014	1.40121E-03	1.99805E-02	7014	14.0033	NITROGEN-14 ENDF/B-IV MAT 1275	UPDATED
08/12/94						
8008016	2.60749E-02	4.24574E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
8013027	7.78110E-03	2.13789E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE = 9 DENSITY(G/CC) = 0.59890						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
9001001	4.00615E-02	1.11927E-01	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
9008016	2.00308E-02	8.88074E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
3001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
8001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
9001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
6005010					B-10 1273 218NGP 042375 P-3 293K	UPDATED 08/12/94

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

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8005010 B-10 1273 218NGP 042375 P-3 293K      UPDATED 08/12/94
6005011 BORON-11      ENDF/B-IV MAT 1160      UPDATED 08/12/94
8005011 BORON-11      ENDF/B-IV MAT 1160      UPDATED 08/12/94
6006012 CARBON-12      ENDF/B-IV MAT 1274/THRM1065    UPDATED 08/12/94
8006012 CARBON-12      ENDF/B-IV MAT 1274/THRM1065    UPDATED 08/12/94
8007014 NITROGEN-14     ENDF/B-IV MAT 1275      UPDATED 08/12/94
1008016 OXYGEN-16         ENDF/B-IV MAT 1276      UPDATED 08/12/94
3008016 OXYGEN-16         ENDF/B-IV MAT 1276      UPDATED 08/12/94
8008016 OXYGEN-16         ENDF/B-IV MAT 1276      UPDATED 08/12/94
9008016 OXYGEN-16         ENDF/B-IV MAT 1276      UPDATED 08/12/94
4013027 AL-27 1193 218 GP 040375(5)      UPDATED 08/12/94
6013027 AL-27 1193 218 GP 040375(5)      UPDATED 08/12/94
8013027 AL-27 1193 218 GP 040375(5)      UPDATED 08/12/94
5024304 CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)  UPDATED 08/12/94
5025055 MANGANESE-55      ENDF/B-IV MAT 1197      UPDATED 08/12/94
5026304 FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)  UPDATED 08/12/94
5028304 NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)  UPDATED 08/12/94
2040302 ZIRCALLOY         ENDF/B-IV MAT 1284      UPDATED 08/12/94
7082000 PB 1288 218NGP 042375 P-3 293K      UPDATED 08/12/94
1092235 URANIUM-235     ENDF/B-IV MAT 1261      UPDATED 08/12/94
1092238 URANIUM-238     ENDF/B-IV MAT 1262      UPDATED 08/12/94

..... 0 IO'S WERE USED MIXING CROSS-SECTIONS .....

1-D CROSS SECTION ARRAY ID NUMBERS
1 2002 1452 27 18 1018

..... 0 IO'S WERE USED PREPARING THE CROSS SECTIONS .....

*****
***
*** TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0. ***
***
*****
***** ADDITIONAL INFORMATION *****
***
*** NUMBER OF ENERGY GROUPS 27 USE LATTICE GEOMETRY YES ***
***
*** NO. OF FISSION SPECTRUM SOURCE GROUP 1 GLOBAL ARRAY NUMBER 60 ***
***
*** NO. OF SCATTERING ANGLES IN XSECS 2 NUMBER OF UNITS IN THE GLOBAL X DIR. 1 ***
***
*** ENTRIES/NEUTRON IN THE NEUTRON BANK 33 NUMBER OF UNITS IN THE GLOBAL Y DIR. 1 ***
***
*** ENTRIES/NEUTRON IN THE FISSION BANK 26 NUMBER OF UNITS IN THE GLOBAL Z DIR. 4 ***
***

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Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

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*** NUMBER OF MIXTURES USED          9      USE A GLOBAL REFLECTOR          YES ***
***                                     1      USE NESTED HOLES            YES ***
*** NUMBER OF BIAS ID'S USED          1      USE NESTED HOLES            YES ***
***                                     0      NUMBER OF HOLES              48 ***
*** NUMBER OF DIFFERENTIAL ALBEDOS USED 0      NUMBER OF HOLES              48 ***
*** TOTAL INPUT GEOMETRY REGIONS      136    MAXIMUM HOLE NESTING LEVEL    3 ***
***                                     136    MAXIMUM HOLE NESTING LEVEL    3 ***
*** NUMBER OF GEOMETRY REGIONS USED    136    USE NESTED ARRAYS            YES ***
***                                     136    USE NESTED ARRAYS            YES ***
*** LARGEST GEOMETRY UNIT NUMBER        120    NUMBER OF ARRAYS USED        27 ***
***                                     120    NUMBER OF ARRAYS USED        27 ***
*** LARGEST ARRAY NUMBER                60     MAXIMUM ARRAY NESTING LEVEL   4 ***
***                                     60     MAXIMUM ARRAY NESTING LEVEL   4 ***
*** +X BOUNDARY CONDITION               MIR    -X BOUNDARY CONDITION        MIR ***
***                                     MIR    -X BOUNDARY CONDITION        MIR ***
*** +Y BOUNDARY CONDITION               MIR    -Y BOUNDARY CONDITION        MIR ***
***                                     MIR    -Y BOUNDARY CONDITION        MIR ***
*** +Z BOUNDARY CONDITION               PER    -Z BOUNDARY CONDITION        PER ***
***                                     PER    -Z BOUNDARY CONDITION        PER ***
***                                     PER    -Z BOUNDARY CONDITION        PER ***
*****
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.

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GENERATION	GENERATION K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	494	INDEPENDENT	FISSION POINTS WERE GENERATED		
1	4.48332E-01	3.85500E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	477	INDEPENDENT	FISSION POINTS WERE GENERATED		
2	4.43293E-01	7.19500E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	534	INDEPENDENT	FISSION POINTS WERE GENERATED		
3	4.51460E-01	1.05467E+00	4.51460E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	4.67077E-01	1.39517E+00	4.59269E-01	7.80828E-03	0.00000E+00	0.00000E+00
5	4.55744E-01	1.73850E+00	4.58094E-01	4.65870E-03	0.00000E+00	0.00000E+00
6	4.51320E-01	2.08533E+00	4.56400E-01	3.70404E-03	0.00000E+00	0.00000E+00
7	4.57700E-01	2.43050E+00	4.56660E-01	2.88088E-03	0.00000E+00	0.00000E+00
8	4.67890E-01	2.77283E+00	4.58532E-01	3.00603E-03	0.00000E+00	0.00000E+00
9	4.63798E-01	3.12433E+00	4.59284E-01	2.64962E-03	0.00000E+00	0.00000E+00
10	4.45528E-01	3.46400E+00	4.57565E-01	2.86743E-03	0.00000E+00	0.00000E+00
11	4.61534E-01	3.79817E+00	4.58006E-01	2.56700E-03	0.00000E+00	0.00000E+00
12	4.70475E-01	4.14150E+00	4.59253E-01	2.61275E-03	0.00000E+00	0.00000E+00
13	4.53302E-01	4.47733E+00	4.58712E-01	2.42445E-03	0.00000E+00	0.00000E+00
14	4.40243E-01	4.81433E+00	4.57173E-01	2.69573E-03	0.00000E+00	0.00000E+00
15	4.57565E-01	5.15667E+00	4.57203E-01	2.47989E-03	0.00000E+00	0.00000E+00
16	4.61519E-01	5.50267E+00	4.57511E-01	2.31654E-03	0.00000E+00	0.00000E+00
17	4.64866E-01	5.84050E+00	4.58001E-01	2.21162E-03	0.00000E+00	0.00000E+00
18	4.69900E-01	6.19100E+00	4.58745E-01	2.19839E-03	0.00000E+00	0.00000E+00
19	4.58153E-01	6.52983E+00	4.58710E-01	2.06532E-03	0.00000E+00	0.00000E+00
20	4.61944E-01	6.87117E+00	4.58890E-01	1.95547E-03	0.00000E+00	0.00000E+00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

21	4.59987E-01	7.20717E+00	4.58948E-01	1.85059E-03	0.00000E+00	0.00000E+00
22	4.51837E-01	7.55317E+00	4.58592E-01	1.79126E-03	0.00000E+00	0.00000E+00
23	4.57642E-01	7.89467E+00	4.58547E-01	1.70443E-03	0.00000E+00	0.00000E+00
24	4.41768E-01	8.24533E+00	4.57784E-01	1.79517E-03	0.00000E+00	0.00000E+00
25	4.50166E-01	8.58950E+00	4.57453E-01	1.74703E-03	0.00000E+00	0.00000E+00
26	4.82646E-01	8.93367E+00	4.58503E-01	1.97476E-03	0.00000E+00	0.00000E+00
27	4.68136E-01	9.28067E+00	4.58888E-01	1.93292E-03	0.00000E+00	0.00000E+00
28	4.56009E-01	9.62200E+00	4.58777E-01	1.86039E-03	0.00000E+00	0.00000E+00
29	4.56370E-01	9.95800E+00	4.58688E-01	1.79238E-03	0.00000E+00	0.00000E+00
30	4.55737E-01	1.03105E+01	4.58583E-01	1.73039E-03	0.00000E+00	0.00000E+00
31	4.59289E-01	1.06557E+01	4.58607E-01	1.66984E-03	0.00000E+00	0.00000E+00
32	4.50980E-01	1.09878E+01	4.58353E-01	1.63312E-03	0.00000E+00	0.00000E+00
33	4.53890E-01	1.13348E+01	4.58209E-01	1.58611E-03	0.00000E+00	0.00000E+00
34	4.61455E-01	1.16790E+01	4.58310E-01	1.53909E-03	0.00000E+00	0.00000E+00
35	4.45631E-01	1.20068E+01	4.57926E-01	1.54041E-03	0.00000E+00	0.00000E+00
36	4.61928E-01	1.23408E+01	4.58044E-01	1.49905E-03	0.00000E+00	0.00000E+00
37	4.65105E-01	1.26797E+01	4.58246E-01	1.46950E-03	0.00000E+00	0.00000E+00
38	4.50275E-01	1.30202E+01	4.58024E-01	1.44516E-03	0.00000E+00	0.00000E+00
39	4.51914E-01	1.33652E+01	4.57859E-01	1.41523E-03	0.00000E+00	0.00000E+00
40	4.66734E-01	1.37077E+01	4.58093E-01	1.39714E-03	0.00000E+00	0.00000E+00
41	4.53626E-01	1.40500E+01	4.57978E-01	1.36565E-03	0.00000E+00	0.00000E+00
42	4.45578E-01	1.43942E+01	4.57668E-01	1.36669E-03	0.00000E+00	0.00000E+00
43	4.59431E-01	1.47430E+01	4.57711E-01	1.33364E-03	0.00000E+00	0.00000E+00
44	4.68701E-01	1.50898E+01	4.57973E-01	1.32754E-03	0.00000E+00	0.00000E+00
45	4.61495E-01	1.54313E+01	4.58055E-01	1.29889E-03	0.00000E+00	0.00000E+00
46	4.44256E-01	1.57773E+01	4.57741E-01	1.30720E-03	0.00000E+00	0.00000E+00
47	4.47966E-01	1.61115E+01	4.57524E-01	1.29615E-03	0.00000E+00	0.00000E+00
48	4.55078E-01	1.64502E+01	4.57471E-01	1.26878E-03	0.00000E+00	0.00000E+00
49	4.57880E-01	1.67862E+01	4.57479E-01	1.24152E-03	0.00000E+00	0.00000E+00
50	4.66583E-01	1.71303E+01	4.57669E-01	1.23009E-03	0.00000E+00	0.00000E+00
51	4.54401E-01	1.74692E+01	4.57602E-01	1.20656E-03	0.00000E+00	0.00000E+00
52	4.65092E-01	1.78160E+01	4.57752E-01	1.19164E-03	0.00000E+00	0.00000E+00
53	4.56241E-01	1.81685E+01	4.57722E-01	1.16842E-03	0.00000E+00	0.00000E+00
54	4.60041E-01	1.85090E+01	4.57767E-01	1.14659E-03	0.00000E+00	0.00000E+00
55	4.73428E-01	1.88587E+01	4.58063E-01	1.16292E-03	0.00000E+00	0.00000E+00
56	4.72201E-01	1.92002E+01	4.58324E-01	1.17083E-03	0.00000E+00	0.00000E+00
57	4.33760E-01	1.95278E+01	4.57878E-01	1.23307E-03	0.00000E+00	0.00000E+00
58	4.54376E-01	1.98675E+01	4.57815E-01	1.21247E-03	0.00000E+00	0.00000E+00
59	4.64330E-01	2.02080E+01	4.57930E-01	1.19648E-03	0.00000E+00	0.00000E+00
60	4.61638E-01	2.05503E+01	4.57993E-01	1.17740E-03	0.00000E+00	0.00000E+00
61	4.68476E-01	2.08965E+01	4.58171E-01	1.17084E-03	0.00000E+00	0.00000E+00
62	4.61480E-01	2.12378E+01	4.58226E-01	1.15248E-03	0.00000E+00	0.00000E+00
63	4.57134E-01	2.15748E+01	4.58208E-01	1.13357E-03	0.00000E+00	0.00000E+00
64	4.69332E-01	2.19190E+01	4.58388E-01	1.12947E-03	0.00000E+00	0.00000E+00
65	4.43979E-01	2.22577E+01	4.58159E-01	1.13469E-03	0.00000E+00	0.00000E+00
66	4.59447E-01	2.26000E+01	4.58179E-01	1.11700E-03	0.00000E+00	0.00000E+00
67	4.46618E-01	2.29360E+01	4.58001E-01	1.11397E-03	0.00000E+00	0.00000E+00
68	4.48726E-01	2.32775E+01	4.57861E-01	1.10593E-03	0.00000E+00	0.00000E+00
69	4.54201E-01	2.36207E+01	4.57806E-01	1.09067E-03	0.00000E+00	0.00000E+00
70	4.58805E-01	2.39622E+01	4.57821E-01	1.07461E-03	0.00000E+00	0.00000E+00
71	4.48634E-01	2.43137E+01	4.57688E-01	1.06726E-03	0.00000E+00	0.00000E+00
72	4.60830E-01	2.46515E+01	4.57733E-01	1.05286E-03	0.00000E+00	0.00000E+00
73	4.62452E-01	2.49938E+01	4.57799E-01	1.04005E-03	0.00000E+00	0.00000E+00
74	4.58161E-01	2.53490E+01	4.57804E-01	1.02551E-03	0.00000E+00	0.00000E+00
75	4.58480E-01	2.56942E+01	4.57813E-01	1.01141E-03	0.00000E+00	0.00000E+00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

76	4.67372E-01	2.60365E+01	4.57943E-01	1.00598E-03	0.00000E+00	0.00000E+00
77	4.58155E-01	2.63917E+01	4.57945E-01	9.92477E-04	0.00000E+00	0.00000E+00
78	4.34600E-01	2.67258E+01	4.57638E-01	1.02637E-03	0.00000E+00	0.00000E+00
79	4.50901E-01	2.70792E+01	4.57551E-01	1.01673E-03	0.00000E+00	0.00000E+00
80	4.64665E-01	2.74243E+01	4.57642E-01	1.00775E-03	0.00000E+00	0.00000E+00
81	4.59790E-01	2.77703E+01	4.57669E-01	9.95279E-04	0.00000E+00	0.00000E+00
82	4.55906E-01	2.81082E+01	4.57647E-01	9.83006E-04	0.00000E+00	0.00000E+00
83	4.65838E-01	2.84542E+01	4.57748E-01	9.76047E-04	0.00000E+00	0.00000E+00
84	4.69348E-01	2.88075E+01	4.57890E-01	9.74393E-04	0.00000E+00	0.00000E+00
85	4.61978E-01	2.91472E+01	4.57939E-01	9.63841E-04	0.00000E+00	0.00000E+00
86	4.66666E-01	2.94868E+01	4.58043E-01	9.57948E-04	0.00000E+00	0.00000E+00
87	4.58813E-01	2.98300E+01	4.58052E-01	9.46655E-04	0.00000E+00	0.00000E+00
88	4.66602E-01	3.01743E+01	4.58151E-01	9.40850E-04	0.00000E+00	0.00000E+00
89	4.56168E-01	3.05083E+01	4.58128E-01	9.30252E-04	0.00000E+00	0.00000E+00
90	4.62012E-01	3.08490E+01	4.58173E-01	9.20678E-04	0.00000E+00	0.00000E+00
91	4.61616E-01	3.11867E+01	4.58211E-01	9.11097E-04	0.00000E+00	0.00000E+00
92	4.42385E-01	3.15282E+01	4.58035E-01	9.17919E-04	0.00000E+00	0.00000E+00
93	4.45441E-01	3.18705E+01	4.57897E-01	9.18265E-04	0.00000E+00	0.00000E+00
94	4.42622E-01	3.22075E+01	4.57731E-01	9.23280E-04	0.00000E+00	0.00000E+00
95	4.48923E-01	3.25360E+01	4.57636E-01	9.18196E-04	0.00000E+00	0.00000E+00
96	4.69889E-01	3.28830E+01	4.57767E-01	9.17680E-04	0.00000E+00	0.00000E+00
97	4.65123E-01	3.32318E+01	4.57844E-01	9.11264E-04	0.00000E+00	0.00000E+00
98	4.68931E-01	3.35797E+01	4.57960E-01	9.09088E-04	0.00000E+00	0.00000E+00
99	4.32937E-01	3.39102E+01	4.57702E-01	9.35920E-04	0.00000E+00	0.00000E+00
100	4.75083E-01	3.42525E+01	4.57879E-01	9.43147E-04	0.00000E+00	0.00000E+00
101	4.51684E-01	3.45930E+01	4.57816E-01	9.35667E-04	0.00000E+00	0.00000E+00
102	4.76249E-01	3.49455E+01	4.58001E-01	9.44425E-04	0.00000E+00	0.00000E+00
103	4.43920E-01	3.52805E+01	4.57861E-01	9.45364E-04	0.00000E+00	0.00000E+00
104	4.65394E-01	3.56302E+01	4.57935E-01	9.38958E-04	0.00000E+00	0.00000E+00
105	4.56583E-01	3.59707E+01	4.57922E-01	9.29890E-04	0.00000E+00	0.00000E+00
106	4.58477E-01	3.63205E+01	4.57927E-01	9.20921E-04	0.00000E+00	0.00000E+00
107	4.46813E-01	3.66582E+01	4.57822E-01	9.18230E-04	0.00000E+00	0.00000E+00
108	4.61466E-01	3.69933E+01	4.57856E-01	9.10176E-04	0.00000E+00	0.00000E+00
109	4.56823E-01	3.73412E+01	4.57846E-01	9.01681E-04	0.00000E+00	0.00000E+00
110	4.64921E-01	3.76817E+01	4.57912E-01	8.95692E-04	0.00000E+00	0.00000E+00
111	4.68324E-01	3.80305E+01	4.58007E-01	8.92563E-04	0.00000E+00	0.00000E+00
112	4.64874E-01	3.83655E+01	4.58070E-01	8.86612E-04	0.00000E+00	0.00000E+00
113	4.56108E-01	3.87143E+01	4.58052E-01	8.78766E-04	0.00000E+00	0.00000E+00
114	4.60073E-01	3.90640E+01	4.58070E-01	8.71071E-04	0.00000E+00	0.00000E+00
115	4.67658E-01	3.94118E+01	4.58155E-01	8.67488E-04	0.00000E+00	0.00000E+00
116	4.45389E-01	3.97478E+01	4.58043E-01	8.67106E-04	0.00000E+00	0.00000E+00
117	4.61144E-01	4.01012E+01	4.58070E-01	8.59956E-04	0.00000E+00	0.00000E+00
118	4.67560E-01	4.04380E+01	4.58152E-01	8.56426E-04	0.00000E+00	0.00000E+00
119	4.53482E-01	4.07840E+01	4.58112E-01	8.50012E-04	0.00000E+00	0.00000E+00
120	4.66085E-01	4.11273E+01	4.58179E-01	8.45483E-04	0.00000E+00	0.00000E+00
121	4.57045E-01	4.14623E+01	4.58170E-01	8.38402E-04	0.00000E+00	0.00000E+00
122	4.58942E-01	4.18112E+01	4.58176E-01	8.31411E-04	0.00000E+00	0.00000E+00
123	4.54475E-01	4.21535E+01	4.58146E-01	8.25078E-04	0.00000E+00	0.00000E+00
124	4.49265E-01	4.24932E+01	4.58073E-01	8.21519E-04	0.00000E+00	0.00000E+00
125	4.78256E-01	4.28447E+01	4.58237E-01	8.31171E-04	0.00000E+00	0.00000E+00
126	4.58493E-01	4.31962E+01	4.58239E-01	8.24443E-04	0.00000E+00	0.00000E+00
127	4.63676E-01	4.35458E+01	4.58283E-01	8.18977E-04	0.00000E+00	0.00000E+00
128	4.52209E-01	4.38865E+01	4.58234E-01	8.13879E-04	0.00000E+00	0.00000E+00
129	4.77728E-01	4.42352E+01	4.58388E-01	8.21906E-04	0.00000E+00	0.00000E+00
130	4.59576E-01	4.45895E+01	4.58397E-01	8.15512E-04	0.00000E+00	0.00000E+00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

131	4.57866E-01	4.49337E+01	4.58393E-01	8.09176E-04	0.00000E+00	0.00000E+00
132	4.53502E-01	4.52733E+01	4.58355E-01	8.03808E-04	0.00000E+00	0.00000E+00
133	4.45038E-01	4.56128E+01	4.58254E-01	8.04101E-04	0.00000E+00	0.00000E+00
134	4.51782E-01	4.59590E+01	4.58205E-01	7.99490E-04	0.00000E+00	0.00000E+00
135	4.60021E-01	4.62948E+01	4.58218E-01	7.93574E-04	0.00000E+00	0.00000E+00
136	4.66765E-01	4.66465E+01	4.58282E-01	7.90207E-04	0.00000E+00	0.00000E+00
137	4.63688E-01	4.69907E+01	4.58322E-01	7.85353E-04	0.00000E+00	0.00000E+00
138	4.49183E-01	4.73285E+01	4.58255E-01	7.82448E-04	0.00000E+00	0.00000E+00
139	4.57541E-01	4.76772E+01	4.58250E-01	7.76734E-04	0.00000E+00	0.00000E+00
140	4.63069E-01	4.80250E+01	4.58285E-01	7.71875E-04	0.00000E+00	0.00000E+00
141	4.60385E-01	4.83683E+01	4.58300E-01	7.66451E-04	0.00000E+00	0.00000E+00
142	4.68164E-01	4.87190E+01	4.58370E-01	7.64212E-04	0.00000E+00	0.00000E+00
143	4.66874E-01	4.90695E+01	4.58431E-01	7.61165E-04	0.00000E+00	0.00000E+00
144	4.57725E-01	4.94092E+01	4.58426E-01	7.55802E-04	0.00000E+00	0.00000E+00
145	4.56155E-01	4.97525E+01	4.58410E-01	7.50666E-04	0.00000E+00	0.00000E+00
146	4.60930E-01	5.01022E+01	4.58427E-01	7.45641E-04	0.00000E+00	0.00000E+00
147	4.48704E-01	5.04390E+01	4.58360E-01	7.43511E-04	0.00000E+00	0.00000E+00
148	4.49969E-01	5.07787E+01	4.58303E-01	7.40634E-04	0.00000E+00	0.00000E+00
149	4.55845E-01	5.11183E+01	4.58286E-01	7.35769E-04	0.00000E+00	0.00000E+00
150	4.49922E-01	5.14633E+01	4.58229E-01	7.32962E-04	0.00000E+00	0.00000E+00
151	4.44525E-01	5.18048E+01	4.58138E-01	7.33813E-04	0.00000E+00	0.00000E+00
152	4.49503E-01	5.21362E+01	4.58080E-01	7.31174E-04	0.00000E+00	0.00000E+00
153	4.57337E-01	5.24758E+01	4.58075E-01	7.26332E-04	0.00000E+00	0.00000E+00
154	4.52312E-01	5.28063E+01	4.58037E-01	7.22534E-04	0.00000E+00	0.00000E+00
155	4.51315E-01	5.31468E+01	4.57993E-01	7.19139E-04	0.00000E+00	0.00000E+00
156	4.70381E-01	5.34938E+01	4.58074E-01	7.18968E-04	0.00000E+00	0.00000E+00
157	4.51343E-01	5.38380E+01	4.58030E-01	7.15633E-04	0.00000E+00	0.00000E+00
158	4.51194E-01	5.41758E+01	4.57986E-01	7.12380E-04	0.00000E+00	0.00000E+00
159	4.58894E-01	5.45337E+01	4.57992E-01	7.07852E-04	0.00000E+00	0.00000E+00
160	4.63779E-01	5.48770E+01	4.58029E-01	7.04311E-04	0.00000E+00	0.00000E+00
161	4.72445E-01	5.52340E+01	4.58119E-01	7.05715E-04	0.00000E+00	0.00000E+00
162	4.48802E-01	5.55672E+01	4.58061E-01	7.03705E-04	0.00000E+00	0.00000E+00
163	4.57398E-01	5.59087E+01	4.58057E-01	6.99332E-04	0.00000E+00	0.00000E+00
164	4.59861E-01	5.62483E+01	4.58068E-01	6.95091E-04	0.00000E+00	0.00000E+00
165	4.47229E-01	5.65815E+01	4.58002E-01	6.94007E-04	0.00000E+00	0.00000E+00
166	4.52014E-01	5.69183E+01	4.57965E-01	6.90728E-04	0.00000E+00	0.00000E+00
167	4.57747E-01	5.72498E+01	4.57964E-01	6.86530E-04	0.00000E+00	0.00000E+00
168	4.73331E-01	5.76050E+01	4.58056E-01	6.88632E-04	0.00000E+00	0.00000E+00
169	4.56556E-01	5.79520E+01	4.58047E-01	6.84555E-04	0.00000E+00	0.00000E+00
170	4.38781E-01	5.82897E+01	4.57933E-01	6.90064E-04	0.00000E+00	0.00000E+00
171	4.66030E-01	5.86367E+01	4.57981E-01	6.87640E-04	0.00000E+00	0.00000E+00
172	4.50684E-01	5.89753E+01	4.57938E-01	6.84929E-04	0.00000E+00	0.00000E+00
173	4.63556E-01	5.93242E+01	4.57971E-01	6.81704E-04	0.00000E+00	0.00000E+00
174	4.55825E-01	5.96583E+01	4.57958E-01	6.77844E-04	0.00000E+00	0.00000E+00
175	4.49008E-01	5.99952E+01	4.57906E-01	6.75898E-04	0.00000E+00	0.00000E+00
176	4.52091E-01	6.03338E+01	4.57873E-01	6.72832E-04	0.00000E+00	0.00000E+00
177	4.64346E-01	6.06863E+01	4.57910E-01	6.69999E-04	0.00000E+00	0.00000E+00
178	4.59222E-01	6.10313E+01	4.57917E-01	6.66223E-04	0.00000E+00	0.00000E+00
179	4.52591E-01	6.13738E+01	4.57887E-01	6.63131E-04	0.00000E+00	0.00000E+00
180	4.57701E-01	6.17180E+01	4.57886E-01	6.59396E-04	0.00000E+00	0.00000E+00
181	4.66480E-01	6.20687E+01	4.57934E-01	6.57457E-04	0.00000E+00	0.00000E+00
182	4.42807E-01	6.24008E+01	4.57850E-01	6.59173E-04	0.00000E+00	0.00000E+00
183	4.44915E-01	6.27470E+01	4.57779E-01	6.59405E-04	0.00000E+00	0.00000E+00
184	4.63822E-01	6.30893E+01	4.57812E-01	6.56613E-04	0.00000E+00	0.00000E+00
185	4.54129E-01	6.34280E+01	4.57792E-01	6.53325E-04	0.00000E+00	0.00000E+00

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

186	4.60407E-01	6.37667E+01	4.57806E-01	6.49920E-04	0.00000E+00	0.00000E+00
187	4.52227E-01	6.41145E+01	4.57776E-01	6.47100E-04	0.00000E+00	0.00000E+00
188	4.77262E-01	6.44597E+01	4.57881E-01	6.52082E-04	0.00000E+00	0.00000E+00
189	4.72561E-01	6.48140E+01	4.57959E-01	6.53320E-04	0.00000E+00	0.00000E+00
190	4.34712E-01	6.51480E+01	4.57836E-01	6.61497E-04	0.00000E+00	0.00000E+00
191	4.51545E-01	6.54877E+01	4.57802E-01	6.58829E-04	0.00000E+00	0.00000E+00
192	4.76042E-01	6.58402E+01	4.57898E-01	6.62346E-04	0.00000E+00	0.00000E+00
193	4.72495E-01	6.61953E+01	4.57975E-01	6.63286E-04	0.00000E+00	0.00000E+00
194	4.47268E-01	6.65340E+01	4.57919E-01	6.62175E-04	0.00000E+00	0.00000E+00
195	4.60527E-01	6.68755E+01	4.57932E-01	6.58874E-04	0.00000E+00	0.00000E+00
196	4.63248E-01	6.72142E+01	4.57960E-01	6.56041E-04	0.00000E+00	0.00000E+00
197	4.55099E-01	6.75565E+01	4.57945E-01	6.52833E-04	0.00000E+00	0.00000E+00
198	4.59402E-01	6.79017E+01	4.57953E-01	6.49536E-04	0.00000E+00	0.00000E+00
199	4.60671E-01	6.82458E+01	4.57966E-01	6.46378E-04	0.00000E+00	0.00000E+00
200	4.59900E-01	6.85882E+01	4.57976E-01	6.43179E-04	0.00000E+00	0.00000E+00
201	4.65762E-01	6.89398E+01	4.58015E-01	6.41134E-04	0.00000E+00	0.00000E+00
202	4.41512E-01	6.92738E+01	4.57933E-01	6.43235E-04	0.00000E+00	0.00000E+00
203	4.63288E-01	6.96208E+01	4.57959E-01	6.40581E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123

EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.

LIFETIME = 9.45512E-06 + OR - 5.08997E-08

GENERATION TIME = 1.85503E-06 + OR - 1.01955E-08

NU BAR = 2.54494E+00 + OR - 4.16575E-04

AVERAGE FISSION GROUP = 6.81874E+00 + OR - 6.06833E-03

ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 6.68650E+04 + OR - 4.50944E+02

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.45799	+ OR - 0.00064	0.45735 TO 0.45863	0.45671 TO 0.45928	0.45606 TO 0.45992	200000
4	0.45795	+ OR - 0.00064	0.45730 TO 0.45859	0.45666 TO 0.45924	0.45601 TO 0.45988	199000
5	0.45796	+ OR - 0.00065	0.45731 TO 0.45861	0.45666 TO 0.45925	0.45601 TO 0.45990	198000
6	0.45799	+ OR - 0.00065	0.45734 TO 0.45864	0.45669 TO 0.45929	0.45604 TO 0.45994	197000
7	0.45799	+ OR - 0.00065	0.45734 TO 0.45865	0.45669 TO 0.45930	0.45603 TO 0.45995	196000
8	0.45794	+ OR - 0.00065	0.45729 TO 0.45860	0.45663 TO 0.45925	0.45598 TO 0.45991	195000
9	0.45791	+ OR - 0.00066	0.45725 TO 0.45857	0.45660 TO 0.45923	0.45594 TO 0.45988	194000
10	0.45798	+ OR - 0.00066	0.45732 TO 0.45863	0.45666 TO 0.45929	0.45600 TO 0.45995	193000
11	0.45796	+ OR - 0.00066	0.45730 TO 0.45862	0.45664 TO 0.45928	0.45597 TO 0.45994	192000
12	0.45789	+ OR - 0.00066	0.45723 TO 0.45855	0.45657 TO 0.45921	0.45591 TO 0.45988	191000

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

17	0.45796	+ OR - 0.00067	0.45729 TO 0.45863	0.45662 TO 0.45930	0.45594 TO 0.45997	186000
22	0.45789	+ OR - 0.00068	0.45720 TO 0.45857	0.45652 TO 0.45926	0.45584 TO 0.45994	181000
27	0.45783	+ OR - 0.00068	0.45715 TO 0.45851	0.45647 TO 0.45919	0.45579 TO 0.45987	176000
32	0.45789	+ OR - 0.00070	0.45719 TO 0.45859	0.45649 TO 0.45929	0.45580 TO 0.45998	171000
37	0.45790	+ OR - 0.00071	0.45719 TO 0.45861	0.45647 TO 0.45933	0.45576 TO 0.46004	166000
42	0.45803	+ OR - 0.00073	0.45731 TO 0.45876	0.45658 TO 0.45948	0.45585 TO 0.46021	161000
47	0.45809	+ OR - 0.00074	0.45735 TO 0.45882	0.45661 TO 0.45956	0.45587 TO 0.46030	156000
52	0.45803	+ OR - 0.00076	0.45727 TO 0.45879	0.45651 TO 0.45954	0.45575 TO 0.46030	151000
57	0.45799	+ OR - 0.00075	0.45724 TO 0.45874	0.45649 TO 0.45949	0.45573 TO 0.46025	146000
62	0.45785	+ OR - 0.00077	0.45707 TO 0.45862	0.45630 TO 0.45939	0.45553 TO 0.46016	141000
67	0.45794	+ OR - 0.00079	0.45715 TO 0.45873	0.45637 TO 0.45951	0.45558 TO 0.46030	136000
72	0.45808	+ OR - 0.00081	0.45727 TO 0.45889	0.45646 TO 0.45970	0.45565 TO 0.46051	131000
77	0.45797	+ OR - 0.00084	0.45713 TO 0.45880	0.45629 TO 0.45964	0.45546 TO 0.46048	126000
82	0.45817	+ OR - 0.00085	0.45732 TO 0.45901	0.45648 TO 0.45986	0.45563 TO 0.46070	121000
87	0.45789	+ OR - 0.00087	0.45702 TO 0.45876	0.45615 TO 0.45963	0.45528 TO 0.46050	116000
92	0.45790	+ OR - 0.00089	0.45700 TO 0.45879	0.45611 TO 0.45968	0.45522 TO 0.46058	111000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
97	0.45806	+ OR - 0.00090	0.45716 TO 0.45897	0.45626 TO 0.45987	0.45535 TO 0.46077	106000
102	0.45792	+ OR - 0.00087	0.45705 TO 0.45879	0.45618 TO 0.45966	0.45531 TO 0.46053	101000
107	0.45811	+ OR - 0.00089	0.45722 TO 0.45900	0.45632 TO 0.45990	0.45543 TO 0.46079	96000
112	0.45783	+ OR - 0.00093	0.45690 TO 0.45876	0.45597 TO 0.45968	0.45504 TO 0.46061	91000
117	0.45781	+ OR - 0.00096	0.45685 TO 0.45878	0.45588 TO 0.45974	0.45492 TO 0.46070	86000
122	0.45764	+ OR - 0.00101	0.45663 TO 0.45865	0.45562 TO 0.45966	0.45461 TO 0.46067	81000
127	0.45743	+ OR - 0.00103	0.45640 TO 0.45846	0.45536 TO 0.45949	0.45433 TO 0.46052	76000
132	0.45723	+ OR - 0.00106	0.45617 TO 0.45830	0.45511 TO 0.45936	0.45405 TO 0.46042	71000

Figure 6.7-4 CSAS25 Output for Canistered Yankee Class Fuel - Normal Conditions
(Continued)

137	0.45722	+ OR - 0.00111	0.45611 TO 0.45833	0.45500 TO 0.45944	0.45389 TO 0.46055	66000
142	0.45702	+ OR - 0.00117	0.45584 TO 0.45819	0.45467 TO 0.45937	0.45349 TO 0.46054	61000
147	0.45692	+ OR - 0.00126	0.45566 TO 0.45818	0.45441 TO 0.45944	0.45315 TO 0.46069	56000
152	0.45760	+ OR - 0.00133	0.45627 TO 0.45894	0.45494 TO 0.46027	0.45360 TO 0.46161	51000
157	0.45772	+ OR - 0.00144	0.45629 TO 0.45916	0.45485 TO 0.46059	0.45341 TO 0.46203	46000
162	0.45756	+ OR - 0.00154	0.45602 TO 0.45910	0.45448 TO 0.46064	0.45294 TO 0.46218	41000
167	0.45794	+ OR - 0.00172	0.45622 TO 0.45966	0.45449 TO 0.46139	0.45277 TO 0.46311	36000
172	0.45808	+ OR - 0.00180	0.45628 TO 0.45988	0.45448 TO 0.46168	0.45267 TO 0.46348	31000
177	0.45829	+ OR - 0.00208	0.45621 TO 0.46038	0.45412 TO 0.46246	0.45204 TO 0.46454	26000
182	0.45890	+ OR - 0.00243	0.45647 TO 0.46132	0.45404 TO 0.46375	0.45161 TO 0.46618	21000
187	0.46008	+ OR - 0.00300	0.45708 TO 0.46308	0.45408 TO 0.46609	0.45107 TO 0.46909	16000
192	0.45902	+ OR - 0.00257	0.45644 TO 0.46159	0.45387 TO 0.46416	0.45129 TO 0.46674	11000
197	0.45842	+ OR - 0.00352	0.45490 TO 0.46194	0.45138 TO 0.46546	0.44786 TO 0.46898	6000

RANDOM NUMBER= 51564689044D
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 250.698 CM) (EXT. MOD. VF = 0.

FREQUENCY FOR GENERATIONS 4 TO 203

0.4208 TO 0.4334 *
0.4334 TO 0.4461 *****
0.4461 TO 0.4587 *****
0.4587 TO 0.4714 *****
0.4714 TO 0.4840 *****

FREQUENCY FOR GENERATIONS 54 TO 203

0.4208 TO 0.4334 *
0.4334 TO 0.4461 *****
0.4461 TO 0.4587 *****
0.4587 TO 0.4714 *****
0.4714 TO 0.4840 *****

FREQUENCY FOR GENERATIONS 104 TO 203

0.4208 TO 0.4334 *
0.4334 TO 0.4461 *****
0.4461 TO 0.4587 *****
0.4587 TO 0.4714 *****
0.4714 TO 0.4840 *****

FREQUENCY FOR GENERATIONS 154 TO 203

0.4208 TO 0.4334 *
0.4334 TO 0.4461 *****
0.4461 TO 0.4587 *****
0.4587 TO 0.4714 *****
0.4714 TO 0.4840 *****

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions

```
=CSAS25
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)

THIS IS A MODEL OF THE YNPS NAC-MPC BASKET
LOADED WITH 36 UNITED NUCLEAR TYPE A ASSEMBLIES

PRODUCED FOR THE YANKEE ROWE
STC LICENSE AMENDMENT

INTERIOR MODERATOR VOLUME FRACTION = 1.0
EXTERIOR MODERATOR VOLUME FRACTION = 1.0
CASK TO CASK PITCH = 300 cm
FLOODED PELLET CLAD GAP
NEUTRON SHIELD REMOVED

27GROUPNDF4 LATTICECELL
UO2      1      0.95    293.0 92235 4.0 92238 96.0 END
ZIRCALLOY 2      1.0    293.0      END
H2O      3      1.0    293.0      END
AL       4      1.0    293.0      END
SS304    5      1.0    293.0      END
B-10     6  DEN=2.6226 0.0450 293.0      END
B-11     6  DEN=2.6226 0.2736 293.0      END
C        6  DEN=2.6226 0.0927 293.0      END
AL       6  DEN=2.6226 0.5737 293.0      END
PB       7      1.0    293.0      END
H        8  DEN=1.6291 0.060  293.0      END
O        8  DEN=1.6291 0.425  293.0      END
C        8  DEN=1.6291 0.277  293.0      END
N        8  DEN=1.6291 0.020  293.0      END
AL       8  DEN=1.6291 0.214  293.0      END
B-10     8  DEN=1.6291 0.001  293.0      END
B-11     8  DEN=1.6291 0.004  293.0      END
H2O      9      1.0    293.0      END
H2O     10      1.0    293.0      END
END COMP
SQUAREPITCH 1.1887 0.7887 1 3 0.9271 2 0.8052 10 END
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)
READ PARAM RUN=YES PLT=NO GEN=1003 NPG=1000 TME=500 END PARAM
READ GEOM

WATER LEVEL UNIT CELLS

UNIT 1
COM='FUEL PIN CELL - BETWEEN DISKS'
CYLINDER 1 1 0.3943 2P2.1400
CYLINDER 10 1 0.4026 2P2.1400
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 2
COM='WATER CELL - BETWEEN DISKS'
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 3
COM='DISPLACEMENT CELL - BETWEEN DISKS'
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
UNIT 4
COM='INSTRUMENT TUBE CELL - BETWEEN DISKS'
CYLINDER 3 1 0.4998 2P2.1400
CYLINDER 5 1 0.5442 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
'
' DISK LEVEL UNIT CELLS (BOTH SS AND AL)
'
UNIT 5
COM='FUEL PIN CELL - WITH SS DISK'
CYLINDER 1 1 0.3943 2P0.6604
CYLINDER 10 1 0.4026 2P0.6604
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 6
COM='WATER CELL - WITH SS DISK'
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 7
COM='DISPLACEMENT CELL - WITH SS DISK'
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 8
COM='INSTRUMENT TUBE CELL - WITH SS DISK'
CYLINDER 3 1 0.4998 2P0.6604
CYLINDER 5 1 0.5442 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
'
' WATER LEVEL BORAL SHEETS
'
UNIT 14
COM='X-X BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P9.144 2P0.0318 2P2.1400
CUBOID 4 1 2P9.144 2P0.0953 2P2.1400
UNIT 15
COM='Y-Y BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P0.0318 2P9.144 2P2.1400
CUBOID 4 1 2P0.0953 2P9.144 2P2.1400
'
' DISK LEVEL BORAL SHEETS (AL AND SS)
'
UNIT 16
COM='X-X BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P9.144 2P0.0318 2P0.6604
CUBOID 4 1 2P9.144 2P0.0953 2P0.6604
UNIT 17
COM='Y-Y BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P0.0318 2P9.144 2P0.6604
CUBOID 4 1 2P0.0953 2P9.144 2P0.6604
'
' WATER LEVEL WEB MATERIAL
'
UNIT 20
COM='WATER LEVEL WEB MATERIAL (SMALL) X-X'
CUBOID 3 1 2P10.4635 2P0.9716 2P2.1400
UNIT 21
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
COM='WATER LEVEL WEB MATERIAL (MEDIUM) X-X'
CUBOID 3 1 2P10.4635 2P1.0478 2P2.1400
UNIT 22
COM='WATER LEVEL WEB MATERIAL (LARGE) X-X'
CUBOID 3 1 2P10.4635 2P1.1208 2P2.1400
UNIT 23
COM='WATER LEVEL WEB MATERIAL (LONG) Y-Y'
CUBOID 3 1 2P1.1208 2P79.5249 2P2.1400
'
SUPPORT DISK WEB MATERIAL
'
UNIT 30
COM='SUPPORT DISK WEB MATERIAL (SMALL) X-X'
CUBOID 5 1 2P10.4635 2P0.9716 2P0.6604
UNIT 31
COM='SUPPORT DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 5 1 2P10.4635 2P1.0478 2P0.6604
UNIT 32
COM='SUPPORT DISK WEB MATERIAL (LARGE) X-X'
CUBOID 5 1 2P10.4635 2P1.1208 2P0.6604
UNIT 33
COM='SUPPORT DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 5 1 2P1.1208 2P79.5249 2P0.6604
'
HEAT TRANSFER DISK WEB MATERIAL
'
UNIT 40
COM='HEAT TRANSFER DISK WEB MATERIAL (SMALL) X-X'
CUBOID 4 1 2P10.4635 2P0.9716 2P0.6604
UNIT 41
COM='HEAT TRANSFER DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 4 1 2P10.4635 2P1.0478 2P0.6604
UNIT 42
COM='HEAT TRANSFER DISK WEB MATERIAL (LARGE) X-X'
CUBOID 4 1 2P10.4635 2P1.1208 2P0.6604
UNIT 43
COM='HEAT TRANSFER DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 4 1 2P1.1208 2P79.5249 2P0.6604
'
WATER LEVEL ASSEMBLY ARRAYS
'
UNIT 50
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL'
ARRAY 1 -9.5104 -9.5104 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400
UNIT 51
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P2.1400
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
HOLE 50 0.0 -0.1584 0.0
UNIT 52
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.0 0.1584 0.0
UNIT 53
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 0.0 0.0
UNIT 54
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 0.0 0.0
UNIT 55
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 0.1584 0.0
UNIT 56
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 0.1584 0.0
UNIT 57
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 -0.1584 0.0
UNIT 58
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 -0.1584 0.0
UNIT 59
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P2.1400
SUPPORT DISK LEVEL ASSEMBLY ARRAYS
UNIT 60
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
UNIT 61
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.0 -0.1584 0.0
UNIT 62
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.0 0.1584 0.0
UNIT 63
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
OM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 -0.1584 0.0 0.0
UNIT 64
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 0.0 0.0
UNIT 65
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 0.1584 0.0
UNIT 66
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 -0.1584 0.1584 0.0
UNIT 67
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 -0.1584 0.0
UNIT 68
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 -0.1584 -0.1584 0.0
UNIT 69
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P0.6604
HEAT TRANSFER DISK LEVEL ASSEMBLY ARRAYS
UNIT 70
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441                2P0.6604
CUBOID 5 1 4P10.0661                2P0.6604
CUBOID 3 1 4P10.25681                2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051                2P0.6604
UNIT 71
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.0 -0.1584 0.0
UNIT 72
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.0 0.1584 0.0
UNIT 73
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 -0.1584 0.0 0.0
UNIT 74
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635                2P0.6604
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
HOLE 70 0.1584 0.0 0.0
UNIT 75
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.1584 0.1584 0.0
UNIT 76
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 -0.1584 0.1584 0.0
UNIT 77
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 0.1584 -0.1584 0.0
UNIT 78
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 70 -0.1584 -0.1584 0.0
UNIT 79
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P0.6604
'
' WATER LEVEL BASKET ARRAYS
UNIT 80
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 20 -10.4636 -33.6323 -2.1400
UNIT 81
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 21 -10.4636 -33.6323 -2.1400
UNIT 82
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 22 -10.4636 -56.6549 -2.1400
UNIT 83
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 23 -10.4636 -56.6549 -2.1400
UNIT 84
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY-X)'
ARRAY 24 -10.4636 -79.5251 -2.1400
UNIT 85
COM='13X1 WATER LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 25 -10.4636 -79.5251 -2.1400
UNIT 86
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 26 -10.4636 -79.5251 -2.1400
'
' SUPPORT DISK LEVEL BASKET ARRAYS
UNIT 90
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 30 -10.4636 -33.6323 -0.6604
UNIT 91
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 31 -10.4636 -33.6323 -0.6604
UNIT 92
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
```


Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
ARRAY 32 -10.4636 -56.6549 -0.6604
UNIT 93
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 33 -10.4636 -56.6549 -0.6604
UNIT 94
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 34 -10.4636 -79.5251 -0.6604
UNIT 95
COM='13X1 SUPPORT DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 35 -10.4636 -79.5251 -0.6604
UNIT 96
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 36 -10.4636 -79.5251 -0.6604
'
HEAT TRANSFER DISK LEVEL BASKET ARRAYS
'
UNIT 100
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 40 -10.4636 -33.6323 -0.6604
UNIT 101
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 41 -10.4636 -33.6323 -0.6604
UNIT 102
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 42 -10.4636 -56.6549 -0.6604
UNIT 103
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 43 -10.4636 -56.6549 -0.6604
UNIT 104
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 44 -10.4636 -79.5251 -0.6604
UNIT 105
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 45 -10.4636 -79.5251 -0.6604
UNIT 106
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 46 -10.4636 -79.5251 -0.6604
'
BASKET ARRAY IN TRANSPORT CASK OVERPACK (LEVEL CONSTRUCTION)
'
UNIT 110
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - WATER LEVEL'
ARRAY 50 -33.6323 -79.5251 -2.1400
CYLINDER 3 1 88.1253 2P2.1400
HOLE 80 -69.0614 0.0 0.0
HOLE 82 -46.1912 0.0 0.0
HOLE 81 69.0614 0.0 0.0
HOLE 83 46.1912 0.0 0.0
CYLINDER 5 1 89.7128 2P2.1400
CYLINDER 3 1 90.170 2P2.1400
CYLINDER 5 1 93.98 2P2.1400
CYLINDER 7 1 103.4288 2P2.1400
CYLINDER 5 1 110.109 2P2.1400
CYLINDER 9 1 124.714 2P2.1400
CYLINDER 9 1 125.349 2P2.1400
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
CUBOID 9 1 4P150 2P2.1400
UNIT 111
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - SUPPORT DISK LEVEL'
ARRAY 51 -33.6323 -79.5251 -0.6604
CYLINDER 5 1 87.6046 2P0.6604
HOLE 90 -69.0614 0.0 0.0
HOLE 92 -46.1912 0.0 0.0
HOLE 91 69.0614 0.0 0.0
HOLE 93 46.1912 0.0 0.0
CYLINDER 3 1 88.1253 2P0.6604
CYLINDER 5 1 89.7128 2P0.6604
CYLINDER 3 1 90.170 2P0.6604
CYLINDER 5 1 93.98 2P0.6604
CYLINDER 7 1 103.4288 2P0.6604
CYLINDER 5 1 110.109 2P0.6604
CYLINDER 9 1 124.714 2P0.6604
CYLINDER 9 1 125.349 2P0.6604
CUBOID 9 1 4P150 2P0.6604
UNIT 112
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - HEAT TRANSFER DISK LEVEL'
ARRAY 52 -33.6323 -79.5251 -0.6604
CYLINDER 4 1 87.2490 2P0.6604
HOLE 100 -69.0614 0.0 0.0
HOLE 102 -46.1912 0.0 0.0
HOLE 101 69.0614 0.0 0.0
HOLE 103 46.1912 0.0 0.0
CYLINDER 3 1 88.1253 2P0.6604
CYLINDER 5 1 89.7128 2P0.6604
CYLINDER 3 1 90.170 2P0.6604
CYLINDER 5 1 93.98 2P0.6604
CYLINDER 7 1 103.4288 2P0.6604
CYLINDER 5 1 110.109 2P0.6604
CYLINDER 9 1 124.714 2P0.6604
CYLINDER 9 1 125.349 2P0.6604
CUBOID 9 1 4P150 2P0.6604
GLOBAL UNIT
GLOBAL UNIT 120
ARRAY 60 -175.349 -175.349 0.0
END GEOM
READ ARRAY
ARA=1 NUX=16 NUY=16 NUZ=1 FILL
1 1 1 1 1 1 1 3 2 2 2 2 2 2 2 2
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
3 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 4 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1
END FILL
ARA=2 NUX=16 NUY=16 NUZ=1 FILL
5 5 5 5 5 5 5 5 7 6 6 6 6 6 6 6
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
7 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 8 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
6 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
END FILL
WATER LEVEL ARRAYS
ARA=20 NUX=1 NUY=5 NUZ=1
FILL
55
22
54
22
57
END FILL
ARA=21 NUX=1 NUY=5 NUZ=1
FILL
56
22
53
22
58
END FILL
ARA=22 NUX=1 NUY=9 NUZ=1
FILL
55
21
55
22
54
22
57
21
57
END FILL
ARA=23 NUX=1 NUY=9 NUZ=1
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
FILL
56
21
56
22
53
22
58
21
58
END FILL
ARA=24 NUX=1 NUY=13 NUZ=1
FILL
55
20
55
21
55
22
54
22
57
21
57
20
57
END FILL
ARA=25 NUX=1 NUY=13 NUZ=1
FILL
52
20
52
21
52
22
59
22
51
21
51
20
51
END FILL
ARA=26 NUX=1 NUY=13 NUZ=1
FILL
56
20
56
21
56
22
53
22
58
21
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
58
20
58
END FILL
      SUPPOR DISK LEVEL ARRAYS
ARA=30  NUX=1  NUY=5  NUZ=1
FILL
65
32
64
32
67
END FILL
ARA=31  NUX=1  NUY=5  NUZ=1
FILL
66
32
63
32
68
END FILL
ARA=32  NUX=1  NUY=9  NUZ=1
FILL
65
31
65
32
64
32
67
31
67
END FILL
ARA=33  NUX=1  NUY=9  NUZ=1
FILL
66
31
66
32
63
32
68
31
68
END FILL
ARA=34  NUX=1  NUY=13  NUZ=1
FILL
65
30
65
31
65
32
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
64
32
67
31
67
30
67
END FILL
ARA=35 NUX=1 NUY=13 NUZ=1
FILL
62
30
62
31
62
32
69
32
61
31
61
30
61
END FILL
ARA=36 NUX=1 NUY=13 NUZ=1
FILL
66
30
66
31
66
32
63
32
68
31
68
30
68
END FILL
HEAT TRANSFER DISK LEVEL ARRAYS
ARA=40 NUX=1 NUY=5 NUZ=1
FILL
75
42
74
42
77
END FILL
ARA=41 NUX=1 NUY=5 NUZ=1
FILL
76
42
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
73
42
78
END FILL
ARA=42 NUX=1 NUY=9 NUZ=1
FILL
75
41
75
42
74
42
77
41
77
END FILL
ARA=43 NUX=1 NUY=9 NUZ=1
FILL
76
41
76
42
73
42
78
41
78
END FILL
ARA=44 NUX=1 NUY=13 NUZ=1
FILL
75
40
75
41
75
42
74
42
77
41
77
40
77
END FILL
ARA=45 NUX=1 NUY=13 NUZ=1
FILL
72
40
72
41
72
42
79
42
71
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
41
71
40
71
END FILL
ARA=46 NUX=1 NUY=13 NUZ=1
FILL
76
40
76
41
76
42
73
42
78
41
78
40
78
END FILL
'
' MAJOR ARRAYS
'
ARA=50 NUX=5 NUY=1 NUZ=1
FILL
84 23 85 23 86
END FILL
ARA=51 NUX=5 NUY=1 NUZ=1
FILL
94 33 95 33 96
END FILL
ARA=52 NUX=5 NUY=1 NUZ=1
FILL
104 43 105 43 106
END FILL
'
' GLOBAL ARRAY
'
ARA=60 NUX=1 NUY=1 NUZ=4
FILL
112
110
111
110
END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIR END BOUNDS
READ PLOT
SCR=YES PIC=MAT LPI=10
UAX=1.0 VDN=-1.0 NAX=1500
'
' WHOLE BASKET HORIZONTAL SLICES
'
TTL='BASKET X-Y CROSS SECTION AT Z= 0.635 HEAT TRANSFER DISK LEVEL'
```


Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
XUL= -130 YUL= 130 ZUL= 0.635
XLR= 130 YLR= -130 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 3.44 WATER LEVEL'
XUL= -130 YUL= 130 ZUL= 3.44
XLR= 130 YLR= -130 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 6.236 SS DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 6.236
XLR= 130 YLR= -130 ZLR= 6.236
UAX=1.0 VDN=-1.0 NAX=1500 END
' HEAT TRANSFER DISK LEVEL BASKET QUADRANTS
TTL='BASKET X-Y QUADRANT I HEAT TRANSFER DISK'
XUL= 12. YUL= 80 ZUL= 0.635
XLR= 80.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II HEAT TRANSFER DISK'
XUL= 12.0 YUL= -12.0 ZUL= 0.635
XLR= 80 YLR= -80 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III HEAT TRANSFER DISK'
XUL= -80.0 YUL= -12.0 ZUL= 0.635
XLR= -12.0 YLR= -80.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV HEAT TRANSFER DISK'
XUL= -80.0 YUL= 80.0 ZUL= 0.635
XLR= -12.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
' WATER LEVEL BASKET QUADRANTS
TTL='BASKET X-Y QUADRANT I WATER LEVEL'
XUL= 12. YUL= 80 ZUL= 3.44
XLR= 80.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II WATER LEVEL'
XUL= 12.0 YUL= -12.0 ZUL= 3.44
XLR= 80 YLR= -80 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III WATER LEVEL'
XUL= -80.0 YUL= -12.0 ZUL= 3.44
XLR= -12.0 YLR= -80.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV WATER LEVEL'
XUL= -80.0 YUL= 80.0 ZUL= 3.44
XLR= -12.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
' VERTICAL SLICES
TTL='BASKET X-Z CROSS SECTION ALUMINUM LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 1.27
XLR= 90 YLR=0.4 ZLR= -.1
```

Figure 6.7-5 CSAS25 Input for Canistered Yankee Class Fuel - Accident Conditions
(Continued)

```
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION WATER LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 4.318
XLR= 90 YLR=0.4 ZLR= 1.27
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION SS LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 6.858
XLR= 90 YLR=0.4 ZLR= 5.588
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION ENTIRE MODEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 12
XLR= 90 YLR=0.4 ZLR= 0
UAX=1.0 WDN=-1.0 NAX=1500 END
END PLOT
END DATA
END
```

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
***** PROGRAM: CSAS *****  
*****  
***** CREATION DATE: 03-08-96 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\scale43\exe *****  
*****  
*****  
***** PRODUCTION CODE: CSAS *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAM: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 11/12/96 *****  
*****  
***** TIME OF EXECUTION: 16:53:43 *****  
*****  
*****
```

**** PROBLEM PARAMETERS ****

***** PROBLEM COMPOSITION DESCRIPTION *****

6.7-109

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```

          92238  96.000 WT%
      8016      2.00 ATOMS/MOLECULE
END

SC ZIRCALLOY  STANDARD COMPOSITION
MX           2 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        6.5600 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
          40302      1.00 ATOM/MOLECULE
END

SC H2O        STANDARD COMPOSITION
MX           3 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        0.9982 THEORETICAL DENSITY
NEL          2 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
          1001      2.00 ATOMS/MOLECULE
          8016      1.00 ATOM/MOLECULE
END

SC AL         STANDARD COMPOSITION
MX           4 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        2.7020 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
          13027      1.00 ATOM/MOLECULE
END

SC SS304      STANDARD COMPOSITION
MX           5 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        7.9200 THEORETICAL DENSITY
NEL          4 NO. ELEMENTS
ICP          0 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
          24304      19.000 WT%
          25055      2.000 WT%
          26304      69.500 WT%
          28304      9.500 WT%
END

SC B-10       STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF          0.0450 VOLUME FRACTION
ROTH        2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```

      5010      1.00 ATOM/MOLECULE
END

SC B-11      STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.2736 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
      5011      1.00 ATOM/MOLECULE
END

SC C          STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.0927 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
      6012      1.00 ATOM/MOLECULE
END

SC AL         STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.5737 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
      13027     1.00 ATOM/MOLECULE
END

SC PB         STANDARD COMPOSITION
MX           7 MIXTURE NO.
VF           1.0000 VOLUME FRACTION
ROTH         11.3440 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
      82000     1.00 ATOM/MOLECULE
END

SC H          STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.0600 VOLUME FRACTION
ROTH         1.6291 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
      1001      1.00 ATOM/MOLECULE
END
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```
SC O          STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.4250 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              8016      1.00 ATOM/MOLECULE
END
```

```
SC C          STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.2770 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              6012      1.00 ATOM/MOLECULE
END
```

```
SC N          STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.0200 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              7014      1.00 ATOM/MOLECULE
END
```

```
SC AL         STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.2140 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              13027     1.00 ATOM/MOLECULE
END
```

```
SC B-10       STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.0010 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              5010      1.00 ATOM/MOLECULE
END
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```
SC B-11      STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.0040 VOLUME FRACTION
ROTH         1.6291 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
              5011      1.00 ATOM/MOLECULE
END

SC H2O       STANDARD COMPOSITION
MX           9 MIXTURE NO.
VF           1.0000 VOLUME FRACTION
ROTH         0.9982 THEORETICAL DENSITY
NEL          2 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
              1001      2.00 ATOMS/MOLECULE
              8016      1.00 ATOM/MOLECULE
END

SC H2O       STANDARD COMPOSITION
MX           10 MIXTURE NO.
VF           1.0000 VOLUME FRACTION
ROTH         0.9982 THEORETICAL DENSITY
NEL          2 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
              1001      2.00 ATOMS/MOLECULE
              8016      1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH        1.1887 CM CENTER TO CENTER SPACING
FUELOD       0.7887 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL        1 MIXTURE NO. OF FUEL
MMOD         3 MIXTURE NO. OF MODERATOR
CLADOD       0.9271 CM CLAD OUTER DIAMETER
MCLAD        2 MIXTURE NO. OF CLAD
GAPOD        0.8052 CM GAP OUTER DIAMETER
MGAP         10 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```
*****
*****          PROGRAM VERIFICATION INFORMATION          *****
*****
*****          CODE SYSTEM:  SCALE-PC VERSION:  4.3          *****
*****
*****-----*****
*****
*****          PROGRAM:  000002          *****
*****          CREATION DATE:  09-28-95          *****
*****          VOLUME:  ENG          *****
*****          LIBRARY:  G:\scale43\exe          *****
*****          PRODUCTION CODE:  NITAWL          *****
*****          VERSION:  3.0          *****
*****          JOBNAME:  SCALE-PC          *****
*****          DATE OF EXECUTION:  11/12/96          *****
*****          TIME OF EXECUTION:  16:53:48          *****
*****
*****-----*****
*****
-1Q ARRAY HAS      1 ENTRIES.

0Q ARRAY HAS      9 ENTRIES.

1Q ARRAY HAS     12 ENTRIES.

SELECT 27 NUCLIDES FROM THE MASTER  LIBRARY ON LOGICAL  1
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL  2
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL  3
      TO CREATE THE NEW WORKING LIBRARY ON LOGICAL  4

      4 RESONANCE CALCULATIONS HAVE BEEN REQUESTED
-1 OUTPUT OPTION FOR AMPX FORMATTED CROSS SECTION DATA
2001 MAXIMUM NUMBER OF RESONANCE MESH INTERVALS
      2 ORDER OF RESONANCE LEVEL PROCESSING

THE STORAGE ALLOCATED FOR THIS CASE IS    100000 WORDS

      2Q ARRAY HAS     27 ENTRIES.

      3Q ARRAY HAS     60 ENTRIES.

      4Q ARRAY HAS     27 ENTRIES.
```


Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

GENERAL INFORMATION CONCERNING CROSS SECTION LIBRARY

TAPE IDENTIFICATION NUMBER	4321
NUMBER OF NUCLIDES ON TAPE	27
NUMBER OF NEUTRON ENERGY GROUPS	27
FIRST THERMAL NEUTRON ENERGY GROUP	15
NUMBER OF GAMMA ENERGY GROUPS	0

DIRECT ACCESS UNIT NUMBER 9 REQUIRES 117 BLOCKS OF LENGTH 1680 WORDS
XSDRN TAPE 4321

SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY

BASED ON ENDF-B VERSION 4 DATA

COMPILED FOR NRC 1/27/89

LAST UPDATED

L.M.PETRIE - ORNL

08/12/94

NUCLIDES FROM XSDRN TAPE

1	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001
2	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001
3	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001
4	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	10001001
5	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	6005010
6	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	8005010
7	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	6005011
8	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	8005011
9	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	6006012
10	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	8006012
11	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	8007014
12	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	1008016
13	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	3008016
14	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	8008016
15	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	9008016
16	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	10008016
17	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	4013027
18	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	6013027
19	AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	8013027
20	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5024304
21	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94	5025055
22	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5026304
23	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	5028304
24	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94	2040302
25	PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94	7082000
26	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94	1092235
27	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94	1092238

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	3001001	TEMPERATURE=	293.00
			PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	8001001	TEMPERATURE=	293.00
			PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	9001001	TEMPERATURE=	293.00
			PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94 10001001	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94 6005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94 8005010	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94 6005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94 8005011	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94 6006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94 8006012	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94 8007014	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 1008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 3008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 8008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 9008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94 10008016	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 4013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 6013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94 8013027	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94 5024304	TEMPERATURE=	293.00
		PROCESS NUMBER 1007 IS AT	TEMPERATURE=	293.00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

MANGANESE-55 ENDF/B-IV MAT 1197 UPDATED 08/12/94 5025055 TEMPERATURE= 293.00

GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	54.466	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	2.590	LUMPED NUCLEAR DENSITY	=	1.7363295E-03
SPIN FACTOR (G)	=	14.448	LUMP DIMENSION (A-BAR)	=	0.0000000E+00
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	0.0000000E+00

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 55.845 SIGMA (PER ABSORBER ATOM) = 3.4663022E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 55.925 SIGMA (PER ABSORBER ATOM) = 1.2557598E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-5.518788E-04	0.000000E+00	-3.944190E-01
9	-2.797993E-03	0.000000E+00	-2.293471E+00
10	-3.291452E-01	0.000000E+00	-3.820862E+01
11	-2.680562E+00	0.000000E+00	-1.159996E+02

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	3.33719E+00
FISSION	0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5026304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED 08/12/94 5028304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

ZIRCALLOY ENDF/B-IV MAT 1284 UPDATED 08/12/94 2040302 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	90.436	TEMPERATURE (KELVIN)	=	293.000
-----------------	---	--------	----------------------	---	---------

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

POTENTIAL SCATTER SIGMA = 6.385 LUMPED NUCLEAR DENSITY = 4.3307818E-02
SPIN FACTOR (G) = 1.079 LUMP DIMENSION (A-BAR) = 4.6355000E-01
INNER RADIUS = 4.0259999E-01 DANC OFF CORRECTION (C) = 4.8012701E-01
THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.
VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-3.660543E-04	0.000000E+00	-2.908784E-01
9	-2.489563E-02	0.000000E+00	-1.027892E+00
10	-2.529988E-02	0.000000E+00	-5.793766E-01
11	-8.583183E-02	0.000000E+00	-4.082697E-01

EXCESS RESONANCE INTEGRALS
RESOLVED
ABSORPTION 5.46203E-01
FISSION 0.00000E+00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
PB 1288 218NGP 042375 P-3 293K UPDATED 08/12/94 7082000 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
URANIUM-235 ENDF/B-IV MAT 1261 UPDATED 08/12/94 1092235 TEMPERATURE= 293.00
RESONANCE DATA FOR THIS NUCLIDE
MASS NUMBER (A) = 233.025 TEMPERATURE(KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 11.500 LUMPED NUCLEAR DENSITY = 9.4064139E-04
SPIN FACTOR (G) = 15171.100 LUMP DIMENSION (A-BAR) = 3.9434999E-01
INNER RADIUS = 0.0000000E+00 DANC OFF CORRECTION (C) = 2.9500756E-01
THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
MASS OF MODERATOR-1 = 15.991 SIGMA(PER ABSORBER ATOM)= 1.9199110E+02
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
MASS OF MODERATOR-2 = 238.051 SIGMA(PER ABSORBER ATOM)= 2.9209552E+02
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.
THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.
VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

GROUP	RES ABS	RES FISS	RES SCAT
12	-4.040891E+00	-2.484752E+00	-9.599491E-02
13	-1.260745E+01	-6.154013E+00	-2.681050E-01
14	-9.467398E+00	-5.597835E+00	-6.275433E-02
15	-5.487491E-04	-4.169921E-04	5.016608E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	1.97633E+02
FISSION	1.18625E+02

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-238 ENDF/B-IV MAT 1262

UPDATED 08/12/94 1092238 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	236.006	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	10.599	LUMPED NUCLEAR DENSITY	=	2.2290209E-02
SPIN FACTOR (G)	=	656.527	LUMP DIMENSION (A-BAR)	=	3.9434999E-01
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	2.9500756E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1	=	15.991	SIGMA (PER ABSORBER ATOM)	=	8.1019773E+00
---------------------	---	--------	---------------------------	---	---------------

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2	=	235.044	SIGMA (PER ABSORBER ATOM)	=	5.0228214E-01
---------------------	---	---------	---------------------------	---	---------------

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
9	-4.868848E-02	0.000000E+00	-4.793465E-01
10	-1.166066E+00	-2.527089E-05	-7.025325E+00
11	-9.951198E+00	0.000000E+00	-2.725808E+01
12	-4.311383E+01	0.000000E+00	-5.006018E+01
13	-5.391320E+01	0.000000E+00	-1.768698E+01
14	-1.043868E+02	0.000000E+00	-6.064470E+00
15	-9.692318E-07	0.000000E+00	1.880502E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	1.80430E+01
FISSION	4.90718E-04

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

THIS XSDRN WORKING TAPE WAS CREATED 11/12/96 AT 16:53:48
THE TITLE OF THE PARENT CASE IS AS FOLLOWS
SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
BASED ON ENDF-B VERSION 4 DATA
COMPILED FOR NRC 1/27/89

TAPE ID	4321	NUMBER OF NUCLIDES	27
NUMBER OF NEUTRON GROUPS	27	NUMBER OF GAMMA GROUPS	0
FIRST THERMAL GROUP	15	LOGICAL UNIT	4

TABLE OF CONTENTS

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 3001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 8001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 9001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID 10001001
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 6005010
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 8005010
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	ID 6005011
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	ID 8005011
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	ID 6006012
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	ID 8006012
NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED 08/12/94	ID 8007014
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 1008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 3008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 8008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 9008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 1008016
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 4013027
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 6013027
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 8013027
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5024304
MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED 08/12/94	ID 5025055
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5026304
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED 08/12/94	ID 5028304
ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED 08/12/94	ID 2040302
PB 1288 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID 7082000
URANIUM-235	ENDF/B-IV MAT 1261	UPDATED 08/12/94	ID 1092235
URANIUM-238	ENDF/B-IV MAT 1262	UPDATED 08/12/94	ID 1092238

TAPE COPY USED 0 I/O'S, AND TOOK 0.33 SECONDS

```
*****  
*****  
*****          PROGRAM VERIFICATION INFORMATION          *****  
*****  
*****          CODE SYSTEM:  SCALE-PC VERSION:   4.3      *****  
*****  
*****  
*****  
*****  
*****  
*****  
*****  
*****  
*****  
*****  
*****          PROGRAM:  O00009                          *****  
*****  
*****          CREATION DATE:  03-08-96                  *****  
*****  
*****          VOLUME:    ENG                            *****  
*****  
*****          LIBRARY:   G:\scale43\exe                 *****  
*****  
*****  
*****          PRODUCTION CODE: KENOVA                   *****  
*****  
*****          VERSION:   3.1                             *****  
*****  
*****          JOBNAME:   SCALE-PC                       *****  
*****  
*****          DATE OF EXECUTION:  11/12/96               *****  
*****  
*****          TIME OF EXECUTION:  16:54:00              *****  
*****  
*****  
*****  
*****  
*****
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```

***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)***
***
*****          NUMERIC PARAMETERS          *****
***
***          TME          MAXIMUM PROBLEM TIME (MIN)          500.00          ***
***
***          TBA          TIME PER GENERATION (MIN)          0.50          ***
***
***          GEN          NUMBER OF GENERATIONS          1003          ***
***
***          NPG          NUMBER PER GENERATION          1000          ***
***
***          NSK          NUMBER OF GENERATIONS TO BE SKIPPED          3          ***
***
***          BEG          BEGINNING GENERATION NUMBER          1          ***
***
***          RES          GENERATIONS BETWEEN CHECKPOINTS          0          ***
***
***          X1D          NUMBER OF EXTRA 1-D CROSS SECTIONS          1          ***
***
***          NBK          NEUTRON BANK SIZE          1025          ***
***
***          XNB          EXTRA POSITIONS IN NEUTRON BANK          0          ***
***
***          NFB          FISSION BANK SIZE          1000          ***
***
***          XFB          EXTRA POSITIONS IN FISSION BANK          0          ***
***
***          WTA          DEFAULT VALUE OF WEIGHT AVERAGE          0.5000          ***
***
***          WTH          WEIGHT HIGH FOR SPLITTING          3.0000          ***
***
***          WTL          WEIGHT LOW FOR RUSSIAN ROULETTE          0.3333          ***
***
***          RND          STARTING RANDOM NUMBER          BB827100001          ***
***
***          NBS          NUMBER OF D.A. BLOCKS ON UNIT 8          200          ***
***
***          NLS          LENGTH OF D.A. BLOCKS ON UNIT 8          512          ***
***
***          ADJ          MODE OF CALCULATION          FORWARD          ***
***
***          INPUT DATA WRITTEN ON RESTART UNIT          NO          ***
***
***          BINARY DATA INTERFACE          YES          ***

```


Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```

***
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)***
*****
***          LOGICAL PARAMETERS          ***
***
*** RUN  EXECUTE PROBLEM AFTER CHECKING DATA  YES          PLT  PLOT PICTURE MAP(S)          NO ***
***
*** FLX  COMPUTE FLUX                          NO          FDN  COMPUTE FISSION DENSITIES          NO ***
***
*** SMU  COMPUTE AVG UNIT SELF-MULTIPLICATION  NO          NUB  COMPUTE NU-BAR & AVG FISSION GROUP          YES ***
***
*** MKU  COMPUTE MATRIX K-EFF BY UNIT NUMBER   NO          MKP  COMPUTE MATRIX K-EFF BY UNIT LOCATION          NO ***
***
*** CKU  COMPUTE COFACTOR K-EFF BY UNIT NUMBER NO          CKP  COMPUTE COFACTOR K-EFF BY UNIT LOCATION          NO ***
***
*** FMU  PRINT FISS PROD MATRIX BY UNIT NUMBER NO          FMP  PRINT FISS PROD MATRIX BY UNIT LOCATION          NO ***
***
*** MKH  COMPUTE MATRIX K-EFF BY HOLE NUMBER   NO          MKA  COMPUTE MATRIX K-EFF BY ARRAY NUMBER          NO ***
***
*** CKH  COMPUTE COFACTOR K-EFF BY HOLE NUMBER NO          CKA  COMPUTE COFACTOR K-EFF BY ARRAY NUMBER          NO ***
***
*** FMH  PRINT FISS PROD MATRIX BY HOLE NUMBER NO          FMA  PRINT FISS PROD MATRIX BY ARRAY NUMBER          NO ***
***
*** HHL  COLLECT MATRIX BY HIGHEST HOLE LEVEL  NO          HAL  COLLECT MATRIX BY HIGHEST ARRAY LEVEL          NO ***
***
*** AMX  PRINT ALL MIXED CROSS SECTIONS        NO          FAR  PRINT FIS. AND ABS. BY REGION          NO ***
***
*** XS1  PRINT 1-D MIXTURE X-SECTIONS          NO          GAS  PRINT FAR BY GROUP          NO ***
***
*** XS2  PRINT 2-D MIXTURE X-SECTIONS          NO          PAX  PRINT XSEC-ALBEDO CORRELATION TABLES          NO ***
***
*** XAP  PRINT MIXTURE ANGLES & PROBABILITIES  NO          PWT  PRINT WEIGHT AVERAGE ARRAY          NO ***
***
*** PKI  PRINT FISSION SPECTRUM                NO          PGM  PRINT INPUT GEOMETRY          NO ***
***
*** PID  PRINT EXTRA 1-D CROSS SECTIONS        NO          BUG  PRINT DEBUG INFORMATION          NO ***
***
***                                     TRK  PRINT TRACKING INFORMATION          NO ***

```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

```

*****
PARAMETER INPUT COMPLETED

..... 0 IO'S WERE USED READING THE PARAMETER DATA .....

CROSS SECTIONS READ FROM THE AMPX WORKING LIBRARY ON UNIT 4

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD =3.0E-05

MIXTURE = 1 DENSITY(G/CC) = 10.412
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE
1008016 4.64617E-02 1.18487E-01 8016 15.9904 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED
08/12/94
1092235 9.40641E-04 3.52606E-02 92235 235.0441 URANIUM-235 ENDF/B-IV MAT 1261 UPDATED
08/12/94
1092238 2.22902E-02 8.46253E-01 92238 238.0510 URANIUM-238 ENDF/B-IV MAT 1262 UPDATED
08/12/94

MIXTURE = 2 DENSITY(G/CC) = 6.5600
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE
2040302 4.33078E-02 1.00000E+00 40000 91.2196 ZIRCALLOY ENDF/B-IV MAT 1284 UPDATED
08/12/94

MIXTURE = 3 DENSITY(G/CC) = 0.99817
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE
3001001 6.67692E-02 1.11927E-01 1001 1.0077 HYDROGEN ENDF/B-IV MAT 1269/THRM1002 UPDATED
08/12/94
3008016 3.33846E-02 8.88074E-01 8016 15.9904 OXYGEN-16 ENDF/B-IV MAT 1276 UPDATED
08/12/94

MIXTURE = 4 DENSITY(G/CC) = 2.7020
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE
4013027 6.03066E-02 1.00000E+00 13027 26.9818 AL-27 1193 218 GP 040375(5) UPDATED
08/12/94

MIXTURE = 5 DENSITY(G/CC) = 7.9200
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE
5024304 1.74286E-02 1.90000E-01 24000 51.9957 CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED
08/12/94
5025055 1.73633E-03 1.99999E-02 25055 54.9379 MANGANESE-55 ENDF/B-IV MAT 1197 UPDATED
08/12/94
5026304 5.93579E-02 6.95000E-01 26000 55.8447 FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED
08/12/94
5028304 7.72070E-03 9.50001E-02 28000 58.6872 NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) UPDATED
08/12/94

MIXTURE = 6 DENSITY(G/CC) = 2.5833
NUCLIDE ATOM-DENS. WGT. FRAC. ZA AWT NUCLIDE TITLE

```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

6005010	7.09799E-03	4.56855E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
6005011	3.92499E-02	2.77771E-01	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
6006012	1.22006E-02	9.41116E-02	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
6013027	3.35812E-02	5.82432E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE =	7	DENSITY(G/CC) =	11.344			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K	UPDATED
08/12/94						
MIXTURE =	8	DENSITY(G/CC) =	1.6307			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
8001001	5.84084E-02	5.99323E-02	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
8005010	9.79802E-05	9.99025E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
8005011	3.56450E-04	3.99615E-03	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
8006012	2.26463E-02	2.76729E-01	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
8007014	1.40121E-03	1.99805E-02	7014	14.0033	NITROGEN-14 ENDF/B-IV MAT 1275	UPDATED
08/12/94						
8008016	2.60749E-02	4.24574E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
8013027	7.78110E-03	2.13789E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE =	9	DENSITY(G/CC) =	0.99817			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
9001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
9008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
MIXTURE =	10	DENSITY(G/CC) =	0.99817			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
10001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
10008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
			3001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			8001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			9001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			10001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
			6005010	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
			8005010	B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94
			6005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
			8005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94
			6006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
			8006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

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8007014    NITROGEN-14    ENDF/B-IV MAT 1275    UPDATED 08/12/94
1008016    OXYGEN-16     ENDF/B-IV MAT 1276    UPDATED 08/12/94
3008016    OXYGEN-16     ENDF/B-IV MAT 1276    UPDATED 08/12/94
8008016    OXYGEN-16     ENDF/B-IV MAT 1276    UPDATED 08/12/94
9008016    OXYGEN-16     ENDF/B-IV MAT 1276    UPDATED 08/12/94
10008016   OXYGEN-16     ENDF/B-IV MAT 1276    UPDATED 08/12/94
4013027   AL-27 1193 218 GP 040375(5)    UPDATED 08/12/94
6013027   AL-27 1193 218 GP 040375(5)    UPDATED 08/12/94
8013027   AL-27 1193 218 GP 040375(5)    UPDATED 08/12/94
5024304   CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'    UPDATED 08/12/94
5025055   MANGANESE-55    ENDF/B-IV MAT 1197    UPDATED 08/12/94
5026304   FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'    UPDATED 08/12/94
5028304   NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'    UPDATED 08/12/94
2040302   ZIRCALLOY        ENDF/B-IV MAT 1284    UPDATED 08/12/94
7082000   PB 1288 218NGP 042375 P-3 293K      UPDATED 08/12/94
1092235   URANIUM-235       ENDF/B-IV MAT 1261    UPDATED 08/12/94
1092238   URANIUM-238       ENDF/B-IV MAT 1262    UPDATED 08/12/94

..... 0 IO'S WERE USED MIXING CROSS-SECTIONS .....

1-D CROSS SECTION ARRAY ID NUMBERS
1 2002 1452 27 18 1018

..... 0 IO'S WERE USED PREPARING THE CROSS SECTIONS .....

***
*** TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0) ***
***

***** ADDITIONAL INFORMATION *****

*** NUMBER OF ENERGY GROUPS 27 USE LATTICE GEOMETRY YES ***
*** NO. OF FISSION SPECTRUM SOURCE GROUP 1 GLOBAL ARRAY NUMBER 60 ***
*** NO. OF SCATTERING ANGLES IN XSECS 2 NUMBER OF UNITS IN THE GLOBAL X DIR. 1 ***
*** ENTRIES/NEUTRON IN THE NEUTRON BANK 33 NUMBER OF UNITS IN THE GLOBAL Y DIR. 1 ***
*** ENTRIES/NEUTRON IN THE FISSION BANK 26 NUMBER OF UNITS IN THE GLOBAL Z DIR. 4 ***
*** NUMBER OF MIXTURES USED 9 USE A GLOBAL REFLECTOR YES ***
*** NUMBER OF BIAS ID'S USED 1 USE NESTED HOLES YES ***
*** NUMBER OF DIFFERENTIAL ALBEDOS USED 0 NUMBER OF HOLES 48 ***
*** TOTAL INPUT GEOMETRY REGIONS 136 MAXIMUM HOLE NESTING LEVEL 3 ***
*** NUMBER OF GEOMETRY REGIONS USED 136 USE NESTED ARRAYS YES ***
*** LARGEST GEOMETRY UNIT NUMBER 120 NUMBER OF ARRAYS USED 27 ***

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Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

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*** LARGEST ARRAY NUMBER          60      MAXIMUM ARRAY NESTING LEVEL      4 ***
***
***
*** +X BOUNDARY CONDITION          MIR      -X BOUNDARY CONDITION          MIR ***
***
*** +Y BOUNDARY CONDITION          MIR      -Y BOUNDARY CONDITION          MIR ***
***
*** +Z BOUNDARY CONDITION          PER      -Z BOUNDARY CONDITION          PER ***
***
*****

```

START TYPE 0 WAS USED.

THE NEUTRONS WERE STARTED WITH A FLAT DISTRIBUTION IN A CUBOID DEFINED BY:

+X= 1.24651E+02 -X=-1.75349E+02 +Y= 1.24651E+02 -Y=-1.75349E+02 +Z= 1.12016E+01 -Z= 0.00000E+00
THE FLAG TO START NEUTRONS IN THE REFLECTOR WAS TURNED OFF

0.04183 MINUTES WERE REQUIRED FOR STARTING. TOTAL ELAPSED TIME IS 0.06400 MINUTES.

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

GENERATION	GENERATION K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	920	INDEPENDENT FISSION POINTS WERE GENERATED			
1	8.45260E-01	1.03500E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	948	INDEPENDENT FISSION POINTS WERE GENERATED			
2	8.75995E-01	1.43833E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	952	INDEPENDENT FISSION POINTS WERE GENERATED			
3	8.44310E-01	1.82333E-01	8.44310E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	9.03824E-01	2.21667E-01	8.74067E-01	2.97569E-02	0.00000E+00	0.00000E+00
5	9.25732E-01	2.60000E-01	8.91289E-01	2.43257E-02	0.00000E+00	0.00000E+00
6	8.52908E-01	2.99500E-01	8.81693E-01	1.96961E-02	0.00000E+00	0.00000E+00
7	9.20506E-01	3.39667E-01	8.89456E-01	1.71178E-02	0.00000E+00	0.00000E+00
8	8.74987E-01	3.78167E-01	8.87044E-01	1.41832E-02	0.00000E+00	0.00000E+00
9	9.09645E-01	4.19333E-01	8.90273E-01	1.24142E-02	0.00000E+00	0.00000E+00
10	8.48362E-01	4.58667E-01	8.85034E-01	1.19595E-02	0.00000E+00	0.00000E+00
11	9.01759E-01	4.99000E-01	8.86893E-01	1.07097E-02	0.00000E+00	0.00000E+00
12	8.91590E-01	5.38333E-01	8.87362E-01	9.59060E-03	0.00000E+00	0.00000E+00
13	8.26485E-01	5.77667E-01	8.81828E-01	1.02900E-02	0.00000E+00	0.00000E+00
14	8.67572E-01	6.18833E-01	8.80640E-01	9.46828E-03	0.00000E+00	0.00000E+00
15	8.70696E-01	6.59167E-01	8.79875E-01	8.74308E-03	0.00000E+00	0.00000E+00
16	8.80601E-01	6.97667E-01	8.79927E-01	8.09469E-03	0.00000E+00	0.00000E+00
17	8.89377E-01	7.37000E-01	8.80557E-01	7.56203E-03	0.00000E+00	0.00000E+00
18	8.75835E-01	7.75500E-01	8.80262E-01	7.07979E-03	0.00000E+00	0.00000E+00
19	8.89648E-01	8.14833E-01	8.80814E-01	6.67318E-03	0.00000E+00	0.00000E+00
20	8.78751E-01	8.54167E-01	8.80699E-01	6.29258E-03	0.00000E+00	0.00000E+00
21	8.87345E-01	8.92667E-01	8.81049E-01	5.96245E-03	0.00000E+00	0.00000E+00
22	8.89040E-01	9.32833E-01	8.81449E-01	5.67057E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

23	8.61826E-01	9.73167E-01	8.80514E-01	5.47413E-03	0.00000E+00	0.00000E+00
24	8.76392E-01	1.01350E+00	8.80327E-01	5.22274E-03	0.00000E+00	0.00000E+00
25	8.47984E-01	1.05367E+00	8.78921E-01	5.18483E-03	0.00000E+00	0.00000E+00
26	8.31805E-01	1.09217E+00	8.76958E-01	5.33819E-03	0.00000E+00	0.00000E+00
27	8.80334E-01	1.13150E+00	8.77093E-01	5.12199E-03	0.00000E+00	0.00000E+00
28	8.70337E-01	1.17083E+00	8.76833E-01	4.92791E-03	0.00000E+00	0.00000E+00
29	8.58137E-01	1.20933E+00	8.76140E-01	4.79217E-03	0.00000E+00	0.00000E+00
30	8.61912E-01	1.24967E+00	8.75632E-01	4.64572E-03	0.00000E+00	0.00000E+00
31	8.89233E-01	1.28983E+00	8.76101E-01	4.50713E-03	0.00000E+00	0.00000E+00
32	8.58432E-01	1.33017E+00	8.75512E-01	4.39395E-03	0.00000E+00	0.00000E+00
33	8.98718E-01	1.37050E+00	8.76261E-01	4.31527E-03	0.00000E+00	0.00000E+00
34	8.43359E-01	1.40983E+00	8.75233E-01	4.30289E-03	0.00000E+00	0.00000E+00
35	8.82459E-01	1.44917E+00	8.75452E-01	4.17621E-03	0.00000E+00	0.00000E+00
36	8.73248E-01	1.48950E+00	8.75387E-01	4.05203E-03	0.00000E+00	0.00000E+00
37	8.42016E-01	1.52983E+00	8.74433E-01	4.04844E-03	0.00000E+00	0.00000E+00
38	8.60546E-01	1.57000E+00	8.74048E-01	3.95324E-03	0.00000E+00	0.00000E+00
39	8.86106E-01	1.60933E+00	8.74373E-01	3.85870E-03	0.00000E+00	0.00000E+00
40	8.82359E-01	1.64883E+00	8.74584E-01	3.76166E-03	0.00000E+00	0.00000E+00
41	9.16518E-01	1.68817E+00	8.75659E-01	3.81845E-03	0.00000E+00	0.00000E+00
42	8.75414E-01	1.72833E+00	8.75653E-01	3.72177E-03	0.00000E+00	0.00000E+00
43	8.84311E-01	1.76867E+00	8.75864E-01	3.63599E-03	0.00000E+00	0.00000E+00
44	8.32141E-01	1.80717E+00	8.74823E-01	3.69792E-03	0.00000E+00	0.00000E+00
45	9.00927E-01	1.84650E+00	8.75430E-01	3.66158E-03	0.00000E+00	0.00000E+00
46	8.35074E-01	1.88583E+00	8.74513E-01	3.69309E-03	0.00000E+00	0.00000E+00
47	8.67602E-01	1.92517E+00	8.74359E-01	3.61336E-03	0.00000E+00	0.00000E+00
48	8.73287E-01	1.96450E+00	8.74336E-01	3.53401E-03	0.00000E+00	0.00000E+00
49	8.54860E-01	2.00300E+00	8.73922E-01	3.48274E-03	0.00000E+00	0.00000E+00
50	8.77829E-01	2.04417E+00	8.74003E-01	3.41038E-03	0.00000E+00	0.00000E+00
51	8.55167E-01	2.08550E+00	8.73619E-01	3.36211E-03	0.00000E+00	0.00000E+00
52	8.59312E-01	2.12567E+00	8.73332E-01	3.30658E-03	0.00000E+00	0.00000E+00
53	8.91143E-01	2.16600E+00	8.73682E-01	3.25986E-03	0.00000E+00	0.00000E+00
54	8.96038E-01	2.20633E+00	8.74112E-01	3.22534E-03	0.00000E+00	0.00000E+00
55	8.71498E-01	2.24567E+00	8.74062E-01	3.16428E-03	0.00000E+00	0.00000E+00
56	8.65457E-01	2.28500E+00	8.73903E-01	3.10922E-03	0.00000E+00	0.00000E+00
57	8.91218E-01	2.32533E+00	8.74218E-01	3.06836E-03	0.00000E+00	0.00000E+00
58	8.54651E-01	2.36467E+00	8.73868E-01	3.03326E-03	0.00000E+00	0.00000E+00
59	8.85337E-01	2.40483E+00	8.74069E-01	2.98635E-03	0.00000E+00	0.00000E+00
60	8.80429E-01	2.44517E+00	8.74179E-01	2.93646E-03	0.00000E+00	0.00000E+00
61	9.00667E-01	2.48550E+00	8.74628E-01	2.92097E-03	0.00000E+00	0.00000E+00
62	9.14575E-01	2.52667E+00	8.75294E-01	2.94803E-03	0.00000E+00	0.00000E+00
63	8.76602E-01	2.56700E+00	8.75315E-01	2.89938E-03	0.00000E+00	0.00000E+00
64	8.67834E-01	2.60633E+00	8.75195E-01	2.85478E-03	0.00000E+00	0.00000E+00
65	9.14123E-01	2.64567E+00	8.75813E-01	2.87626E-03	0.00000E+00	0.00000E+00
66	8.65798E-01	2.68600E+00	8.75656E-01	2.83528E-03	0.00000E+00	0.00000E+00
67	8.52804E-01	2.72617E+00	8.75305E-01	2.81338E-03	0.00000E+00	0.00000E+00
68	8.47717E-01	2.76550E+00	8.74887E-01	2.80178E-03	0.00000E+00	0.00000E+00
69	8.91544E-01	2.80500E+00	8.75135E-01	2.77082E-03	0.00000E+00	0.00000E+00
70	9.18735E-01	2.84517E+00	8.75776E-01	2.80406E-03	0.00000E+00	0.00000E+00
71	8.49792E-01	2.88467E+00	8.75400E-01	2.78867E-03	0.00000E+00	0.00000E+00
72	9.00198E-01	2.92300E+00	8.75754E-01	2.77128E-03	0.00000E+00	0.00000E+00
73	9.02772E-01	2.96333E+00	8.76135E-01	2.75834E-03	0.00000E+00	0.00000E+00
74	8.58968E-01	3.00267E+00	8.75896E-01	2.73019E-03	0.00000E+00	0.00000E+00
75	8.81546E-01	3.04200E+00	8.75974E-01	2.69364E-03	0.00000E+00	0.00000E+00
76	8.79696E-01	3.08233E+00	8.76024E-01	2.65747E-03	0.00000E+00	0.00000E+00
77	9.05634E-01	3.12267E+00	8.76419E-01	2.65135E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

78	8.53695E-01	3.16200E+00	8.76120E-01	2.63327E-03	0.00000E+00	0.00000E+00
79	7.91697E-01	3.20133E+00	8.75023E-01	2.82065E-03	0.00000E+00	0.00000E+00
80	8.51839E-01	3.24067E+00	8.74726E-01	2.80007E-03	0.00000E+00	0.00000E+00
81	8.92243E-01	3.28100E+00	8.74948E-01	2.77328E-03	0.00000E+00	0.00000E+00
82	8.30263E-01	3.32033E+00	8.74389E-01	2.79478E-03	0.00000E+00	0.00000E+00
83	8.86753E-01	3.35967E+00	8.74542E-01	2.76428E-03	0.00000E+00	0.00000E+00
84	8.83275E-01	3.39817E+00	8.74648E-01	2.73244E-03	0.00000E+00	0.00000E+00
85	8.65620E-01	3.43750E+00	8.74540E-01	2.70151E-03	0.00000E+00	0.00000E+00
86	8.94753E-01	3.47783E+00	8.74780E-01	2.67998E-03	0.00000E+00	0.00000E+00
87	8.83441E-01	3.51617E+00	8.74882E-01	2.65022E-03	0.00000E+00	0.00000E+00
88	8.99252E-01	3.55567E+00	8.75165E-01	2.63450E-03	0.00000E+00	0.00000E+00
89	8.96617E-01	3.59400E+00	8.75412E-01	2.61569E-03	0.00000E+00	0.00000E+00
90	8.96198E-01	3.63333E+00	8.75648E-01	2.59657E-03	0.00000E+00	0.00000E+00
91	8.35135E-01	3.67367E+00	8.75193E-01	2.60727E-03	0.00000E+00	0.00000E+00
92	8.69912E-01	3.71300E+00	8.75134E-01	2.57881E-03	0.00000E+00	0.00000E+00
93	8.48029E-01	3.75233E+00	8.74836E-01	2.56765E-03	0.00000E+00	0.00000E+00
94	8.53762E-01	3.79183E+00	8.74607E-01	2.54989E-03	0.00000E+00	0.00000E+00
95	9.00097E-01	3.83200E+00	8.74881E-01	2.53717E-03	0.00000E+00	0.00000E+00
96	8.58107E-01	3.87050E+00	8.74703E-01	2.51637E-03	0.00000E+00	0.00000E+00
97	9.35373E-01	3.91083E+00	8.75342E-01	2.57034E-03	0.00000E+00	0.00000E+00
98	8.57543E-01	3.95017E+00	8.75156E-01	2.55018E-03	0.00000E+00	0.00000E+00
99	9.56989E-01	3.99133E+00	8.76000E-01	2.66102E-03	0.00000E+00	0.00000E+00
100	8.58730E-01	4.03067E+00	8.75824E-01	2.63962E-03	0.00000E+00	0.00000E+00
101	8.67060E-01	4.07000E+00	8.75735E-01	2.61432E-03	0.00000E+00	0.00000E+00
102	8.55432E-01	4.10950E+00	8.75532E-01	2.59599E-03	0.00000E+00	0.00000E+00
103	8.67743E-01	4.14967E+00	8.75455E-01	2.57132E-03	0.00000E+00	0.00000E+00
104	8.87242E-01	4.18817E+00	8.75571E-01	2.54861E-03	0.00000E+00	0.00000E+00
105	9.38439E-01	4.22850E+00	8.76181E-01	2.59650E-03	0.00000E+00	0.00000E+00
106	8.70252E-01	4.26967E+00	8.76124E-01	2.57205E-03	0.00000E+00	0.00000E+00
107	8.82507E-01	4.30983E+00	8.76185E-01	2.54816E-03	0.00000E+00	0.00000E+00
108	9.13364E-01	4.34933E+00	8.76535E-01	2.54826E-03	0.00000E+00	0.00000E+00
109	8.98171E-01	4.38867E+00	8.76738E-01	2.53242E-03	0.00000E+00	0.00000E+00
110	9.56280E-01	4.42800E+00	8.77474E-01	2.61473E-03	0.00000E+00	0.00000E+00
111	9.00160E-01	4.46917E+00	8.77682E-01	2.59898E-03	0.00000E+00	0.00000E+00
112	8.48352E-01	4.50950E+00	8.77416E-01	2.58901E-03	0.00000E+00	0.00000E+00
113	8.33516E-01	4.54783E+00	8.77020E-01	2.59588E-03	0.00000E+00	0.00000E+00
114	8.74828E-01	4.58733E+00	8.77001E-01	2.57267E-03	0.00000E+00	0.00000E+00
115	8.72743E-01	4.62567E+00	8.76963E-01	2.55008E-03	0.00000E+00	0.00000E+00
116	8.31336E-01	4.66600E+00	8.76563E-01	2.55911E-03	0.00000E+00	0.00000E+00
117	8.95066E-01	4.70533E+00	8.76724E-01	2.54185E-03	0.00000E+00	0.00000E+00
118	8.41890E-01	4.74567E+00	8.76423E-01	2.53768E-03	0.00000E+00	0.00000E+00
119	8.88804E-01	4.78500E+00	8.76529E-01	2.51812E-03	0.00000E+00	0.00000E+00
120	8.95311E-01	4.82533E+00	8.76688E-01	2.50176E-03	0.00000E+00	0.00000E+00
121	8.68667E-01	4.86367E+00	8.76621E-01	2.48156E-03	0.00000E+00	0.00000E+00
122	8.32125E-01	4.90217E+00	8.76250E-01	2.48857E-03	0.00000E+00	0.00000E+00
123	8.86974E-01	4.94250E+00	8.76339E-01	2.46951E-03	0.00000E+00	0.00000E+00
124	8.71056E-01	4.98183E+00	8.76295E-01	2.44957E-03	0.00000E+00	0.00000E+00
125	9.01007E-01	5.02217E+00	8.76496E-01	2.43786E-03	0.00000E+00	0.00000E+00
126	8.72544E-01	5.06150E+00	8.76464E-01	2.41833E-03	0.00000E+00	0.00000E+00
127	8.84640E-01	5.10167E+00	8.76530E-01	2.39980E-03	0.00000E+00	0.00000E+00
128	8.78852E-01	5.14117E+00	8.76548E-01	2.38075E-03	0.00000E+00	0.00000E+00
129	8.69361E-01	5.18133E+00	8.76492E-01	2.36261E-03	0.00000E+00	0.00000E+00
130	8.94987E-01	5.22250E+00	8.76636E-01	2.34853E-03	0.00000E+00	0.00000E+00
131	9.08787E-01	5.26200E+00	8.76885E-01	2.34354E-03	0.00000E+00	0.00000E+00
132	8.63356E-01	5.30217E+00	8.76781E-01	2.32777E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

133	8.90286E-01	5.34250E+00	8.76884E-01	2.31223E-03	0.00000E+00	0.00000E+00
134	9.04771E-01	5.38283E+00	8.77096E-01	2.30435E-03	0.00000E+00	0.00000E+00
135	9.26312E-01	5.42400E+00	8.77466E-01	2.31671E-03	0.00000E+00	0.00000E+00
136	9.01390E-01	5.46433E+00	8.77644E-01	2.30627E-03	0.00000E+00	0.00000E+00
137	9.32573E-01	5.50550E+00	8.78051E-01	2.32501E-03	0.00000E+00	0.00000E+00
138	8.35606E-01	5.54483E+00	8.77739E-01	2.32885E-03	0.00000E+00	0.00000E+00
139	8.92799E-01	5.58417E+00	8.77849E-01	2.31440E-03	0.00000E+00	0.00000E+00
140	8.79671E-01	5.62267E+00	8.77862E-01	2.29761E-03	0.00000E+00	0.00000E+00
141	8.37348E-01	5.66467E+00	8.77571E-01	2.29957E-03	0.00000E+00	0.00000E+00
142	8.78282E-01	5.70417E+00	8.77576E-01	2.28309E-03	0.00000E+00	0.00000E+00
143	8.88879E-01	5.74533E+00	8.77656E-01	2.26826E-03	0.00000E+00	0.00000E+00
144	8.88779E-01	5.78467E+00	8.77734E-01	2.25359E-03	0.00000E+00	0.00000E+00
145	8.73420E-01	5.82400E+00	8.77704E-01	2.23798E-03	0.00000E+00	0.00000E+00
146	8.87679E-01	5.86333E+00	8.77773E-01	2.22346E-03	0.00000E+00	0.00000E+00
147	8.83991E-01	5.90450E+00	8.77816E-01	2.20849E-03	0.00000E+00	0.00000E+00
148	8.63863E-01	5.94483E+00	8.77721E-01	2.19539E-03	0.00000E+00	0.00000E+00
149	8.91174E-01	5.98517E+00	8.77812E-01	2.18232E-03	0.00000E+00	0.00000E+00
150	8.81947E-01	6.02450E+00	8.77840E-01	2.16771E-03	0.00000E+00	0.00000E+00
151	9.01951E-01	6.06383E+00	8.78002E-01	2.15918E-03	0.00000E+00	0.00000E+00
152	9.16983E-01	6.10417E+00	8.78262E-01	2.16043E-03	0.00000E+00	0.00000E+00
153	8.64412E-01	6.14433E+00	8.78170E-01	2.14803E-03	0.00000E+00	0.00000E+00
154	8.93949E-01	6.18383E+00	8.78274E-01	2.13638E-03	0.00000E+00	0.00000E+00
155	9.01515E-01	6.22317E+00	8.78426E-01	2.12780E-03	0.00000E+00	0.00000E+00
156	8.95357E-01	6.26250E+00	8.78536E-01	2.11679E-03	0.00000E+00	0.00000E+00
157	9.15528E-01	6.30367E+00	8.78774E-01	2.11659E-03	0.00000E+00	0.00000E+00
158	8.58517E-01	6.34400E+00	8.78645E-01	2.10698E-03	0.00000E+00	0.00000E+00
159	9.21723E-01	6.38333E+00	8.78919E-01	2.11142E-03	0.00000E+00	0.00000E+00
160	8.86235E-01	6.42183E+00	8.78965E-01	2.09853E-03	0.00000E+00	0.00000E+00
161	8.56564E-01	6.46200E+00	8.78824E-01	2.09004E-03	0.00000E+00	0.00000E+00
162	9.09129E-01	6.50150E+00	8.79014E-01	2.08556E-03	0.00000E+00	0.00000E+00
163	9.03646E-01	6.54083E+00	8.79167E-01	2.07820E-03	0.00000E+00	0.00000E+00
164	8.81791E-01	6.57917E+00	8.79183E-01	2.06540E-03	0.00000E+00	0.00000E+00
165	8.92339E-01	6.61867E+00	8.79264E-01	2.05427E-03	0.00000E+00	0.00000E+00
166	8.53030E-01	6.65700E+00	8.79104E-01	2.04797E-03	0.00000E+00	0.00000E+00
167	8.79323E-01	6.69633E+00	8.79105E-01	2.03552E-03	0.00000E+00	0.00000E+00
168	8.67184E-01	6.73667E+00	8.79033E-01	2.02449E-03	0.00000E+00	0.00000E+00
169	8.69610E-01	6.77700E+00	8.78977E-01	2.01312E-03	0.00000E+00	0.00000E+00
170	8.55339E-01	6.81733E+00	8.78836E-01	2.00604E-03	0.00000E+00	0.00000E+00
171	8.95872E-01	6.85667E+00	8.78937E-01	1.99669E-03	0.00000E+00	0.00000E+00
172	8.57411E-01	6.89500E+00	8.78810E-01	1.98894E-03	0.00000E+00	0.00000E+00
173	9.01362E-01	6.93533E+00	8.78942E-01	1.98167E-03	0.00000E+00	0.00000E+00
174	8.89882E-01	6.97467E+00	8.79006E-01	1.97114E-03	0.00000E+00	0.00000E+00
175	8.82664E-01	7.01583E+00	8.79027E-01	1.95983E-03	0.00000E+00	0.00000E+00
176	8.97747E-01	7.05617E+00	8.79134E-01	1.95150E-03	0.00000E+00	0.00000E+00
177	8.40187E-01	7.09550E+00	8.78912E-01	1.95304E-03	0.00000E+00	0.00000E+00
178	8.70747E-01	7.13483E+00	8.78866E-01	1.94246E-03	0.00000E+00	0.00000E+00
179	8.71556E-01	7.17433E+00	8.78824E-01	1.93190E-03	0.00000E+00	0.00000E+00
180	9.36237E-01	7.21450E+00	8.79147E-01	1.94790E-03	0.00000E+00	0.00000E+00
181	9.06304E-01	7.25483E+00	8.79299E-01	1.94292E-03	0.00000E+00	0.00000E+00
182	9.06124E-01	7.29517E+00	8.79448E-01	1.93784E-03	0.00000E+00	0.00000E+00
183	8.84520E-01	7.33450E+00	8.79476E-01	1.92731E-03	0.00000E+00	0.00000E+00
184	9.03656E-01	7.37383E+00	8.79608E-01	1.92129E-03	0.00000E+00	0.00000E+00
185	8.52782E-01	7.41317E+00	8.79462E-01	1.91638E-03	0.00000E+00	0.00000E+00
186	8.90962E-01	7.45250E+00	8.79524E-01	1.90696E-03	0.00000E+00	0.00000E+00
187	8.75557E-01	7.49283E+00	8.79503E-01	1.89674E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

188	8.63271E-01	7.53217E+00	8.79416E-01	1.88853E-03	0.00000E+00	0.00000E+00
189	8.61642E-01	7.57250E+00	8.79321E-01	1.88081E-03	0.00000E+00	0.00000E+00
190	8.69610E-01	7.61183E+00	8.79269E-01	1.87149E-03	0.00000E+00	0.00000E+00
191	8.73927E-01	7.65217E+00	8.79241E-01	1.86178E-03	0.00000E+00	0.00000E+00
192	8.84263E-01	7.69233E+00	8.79267E-01	1.85214E-03	0.00000E+00	0.00000E+00
193	8.76541E-01	7.73183E+00	8.79253E-01	1.84248E-03	0.00000E+00	0.00000E+00
194	8.97996E-01	7.77117E+00	8.79350E-01	1.83545E-03	0.00000E+00	0.00000E+00
195	8.88123E-01	7.81133E+00	8.79396E-01	1.82648E-03	0.00000E+00	0.00000E+00
196	8.93078E-01	7.85167E+00	8.79466E-01	1.81841E-03	0.00000E+00	0.00000E+00
197	8.51225E-01	7.89200E+00	8.79322E-01	1.81485E-03	0.00000E+00	0.00000E+00
198	9.15090E-01	7.93217E+00	8.79504E-01	1.81477E-03	0.00000E+00	0.00000E+00
199	9.08755E-01	7.97167E+00	8.79653E-01	1.81163E-03	0.00000E+00	0.00000E+00
200	9.10498E-01	8.01100E+00	8.79808E-01	1.80917E-03	0.00000E+00	0.00000E+00
201	8.87103E-01	8.04933E+00	8.79845E-01	1.80043E-03	0.00000E+00	0.00000E+00
202	8.76015E-01	8.08883E+00	8.79826E-01	1.79151E-03	0.00000E+00	0.00000E+00
203	9.11471E-01	8.12817E+00	8.79983E-01	1.78951E-03	0.00000E+00	0.00000E+00
204	9.27111E-01	8.16667E+00	8.80217E-01	1.79585E-03	0.00000E+00	0.00000E+00
205	8.65983E-01	8.20683E+00	8.80146E-01	1.78836E-03	0.00000E+00	0.00000E+00
206	8.94393E-01	8.24717E+00	8.80216E-01	1.78094E-03	0.00000E+00	0.00000E+00
207	8.80920E-01	8.28750E+00	8.80220E-01	1.77223E-03	0.00000E+00	0.00000E+00
208	8.93639E-01	8.32767E+00	8.80285E-01	1.76481E-03	0.00000E+00	0.00000E+00
209	8.73671E-01	8.36700E+00	8.80253E-01	1.75656E-03	0.00000E+00	0.00000E+00
210	9.03334E-01	8.40650E+00	8.80364E-01	1.75161E-03	0.00000E+00	0.00000E+00
211	8.70822E-01	8.44667E+00	8.80318E-01	1.74381E-03	0.00000E+00	0.00000E+00
212	8.83255E-01	8.48600E+00	8.80332E-01	1.73554E-03	0.00000E+00	0.00000E+00
213	8.78893E-01	8.52633E+00	8.80325E-01	1.72731E-03	0.00000E+00	0.00000E+00
214	9.04516E-01	8.56667E+00	8.80440E-01	1.72292E-03	0.00000E+00	0.00000E+00
215	8.48851E-01	8.60500E+00	8.80291E-01	1.72122E-03	0.00000E+00	0.00000E+00
216	8.59963E-01	8.64450E+00	8.80196E-01	1.71579E-03	0.00000E+00	0.00000E+00
217	9.25021E-01	8.68383E+00	8.80405E-01	1.72047E-03	0.00000E+00	0.00000E+00
218	8.92361E-01	8.72317E+00	8.80460E-01	1.71338E-03	0.00000E+00	0.00000E+00
219	8.69228E-01	8.76350E+00	8.80408E-01	1.70625E-03	0.00000E+00	0.00000E+00
220	9.30453E-01	8.80367E+00	8.80638E-01	1.71385E-03	0.00000E+00	0.00000E+00
221	9.24682E-01	8.84483E+00	8.80839E-01	1.71782E-03	0.00000E+00	0.00000E+00
222	8.86384E-01	8.88333E+00	8.80864E-01	1.71018E-03	0.00000E+00	0.00000E+00
223	8.92452E-01	8.92367E+00	8.80917E-01	1.70323E-03	0.00000E+00	0.00000E+00
224	8.67994E-01	8.96400E+00	8.80858E-01	1.69654E-03	0.00000E+00	0.00000E+00
225	8.68701E-01	9.00417E+00	8.80804E-01	1.68979E-03	0.00000E+00	0.00000E+00
226	8.76143E-01	9.04450E+00	8.80783E-01	1.68236E-03	0.00000E+00	0.00000E+00
227	8.81603E-01	9.08300E+00	8.80787E-01	1.67487E-03	0.00000E+00	0.00000E+00
228	8.77044E-01	9.12317E+00	8.80770E-01	1.66752E-03	0.00000E+00	0.00000E+00
229	8.74798E-01	9.16350E+00	8.80744E-01	1.66037E-03	0.00000E+00	0.00000E+00
230	8.72461E-01	9.20383E+00	8.80708E-01	1.65347E-03	0.00000E+00	0.00000E+00
231	8.75838E-01	9.24400E+00	8.80686E-01	1.64637E-03	0.00000E+00	0.00000E+00
232	8.39798E-01	9.28333E+00	8.80509E-01	1.64881E-03	0.00000E+00	0.00000E+00
233	8.94318E-01	9.32367E+00	8.80568E-01	1.64275E-03	0.00000E+00	0.00000E+00
234	8.44157E-01	9.36400E+00	8.80411E-01	1.64316E-03	0.00000E+00	0.00000E+00
235	9.06152E-01	9.40333E+00	8.80522E-01	1.63982E-03	0.00000E+00	0.00000E+00
236	8.72321E-01	9.44367E+00	8.80487E-01	1.63317E-03	0.00000E+00	0.00000E+00
237	8.54980E-01	9.48383E+00	8.80378E-01	1.62983E-03	0.00000E+00	0.00000E+00
238	9.28608E-01	9.52417E+00	8.80583E-01	1.63572E-03	0.00000E+00	0.00000E+00
239	8.85088E-01	9.56350E+00	8.80602E-01	1.62892E-03	0.00000E+00	0.00000E+00
240	8.35209E-01	9.60283E+00	8.80411E-01	1.63323E-03	0.00000E+00	0.00000E+00
241	9.41049E-01	9.64317E+00	8.80665E-01	1.64606E-03	0.00000E+00	0.00000E+00
242	8.86150E-01	9.68350E+00	8.80687E-01	1.63934E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

243	8.89330E-01	9.72183E+00	8.80723E-01	1.63292E-03	0.00000E+00	0.00000E+00
244	8.48041E-01	9.76033E+00	8.80588E-01	1.63176E-03	0.00000E+00	0.00000E+00
245	8.89503E-01	9.79967E+00	8.80625E-01	1.62544E-03	0.00000E+00	0.00000E+00
246	8.49796E-01	9.83900E+00	8.80499E-01	1.62369E-03	0.00000E+00	0.00000E+00
247	8.93381E-01	9.87850E+00	8.80551E-01	1.61790E-03	0.00000E+00	0.00000E+00
248	9.22387E-01	9.91867E+00	8.80721E-01	1.62026E-03	0.00000E+00	0.00000E+00
249	8.50715E-01	9.95717E+00	8.80600E-01	1.61826E-03	0.00000E+00	0.00000E+00
250	8.74574E-01	9.99833E+00	8.80575E-01	1.61190E-03	0.00000E+00	0.00000E+00
251	9.09363E-01	1.00377E+01	8.80691E-01	1.60957E-03	0.00000E+00	0.00000E+00
252	8.94621E-01	1.00770E+01	8.80747E-01	1.60409E-03	0.00000E+00	0.00000E+00
253	8.83174E-01	1.01173E+01	8.80756E-01	1.59771E-03	0.00000E+00	0.00000E+00
254	8.76024E-01	1.01567E+01	8.80738E-01	1.59147E-03	0.00000E+00	0.00000E+00
255	8.51361E-01	1.01952E+01	8.80622E-01	1.58942E-03	0.00000E+00	0.00000E+00
256	8.65048E-01	1.02345E+01	8.80560E-01	1.58433E-03	0.00000E+00	0.00000E+00
257	8.65416E-01	1.02757E+01	8.80501E-01	1.57923E-03	0.00000E+00	0.00000E+00
258	9.26364E-01	1.03170E+01	8.80680E-01	1.58321E-03	0.00000E+00	0.00000E+00
259	9.24431E-01	1.03563E+01	8.80850E-01	1.58620E-03	0.00000E+00	0.00000E+00
260	8.75630E-01	1.03965E+01	8.80830E-01	1.58017E-03	0.00000E+00	0.00000E+00
261	8.99393E-01	1.04360E+01	8.80902E-01	1.57569E-03	0.00000E+00	0.00000E+00
262	8.62231E-01	1.04762E+01	8.80830E-01	1.57126E-03	0.00000E+00	0.00000E+00
263	8.91752E-01	1.05155E+01	8.80872E-01	1.56579E-03	0.00000E+00	0.00000E+00
264	8.75130E-01	1.05550E+01	8.80850E-01	1.55995E-03	0.00000E+00	0.00000E+00
265	8.99767E-01	1.05962E+01	8.80922E-01	1.55568E-03	0.00000E+00	0.00000E+00
266	8.88867E-01	1.06363E+01	8.80952E-01	1.55006E-03	0.00000E+00	0.00000E+00
267	8.99500E-01	1.06767E+01	8.81022E-01	1.54579E-03	0.00000E+00	0.00000E+00
268	8.67928E-01	1.07170E+01	8.80973E-01	1.54075E-03	0.00000E+00	0.00000E+00
269	8.88036E-01	1.07563E+01	8.80999E-01	1.53520E-03	0.00000E+00	0.00000E+00
270	8.86078E-01	1.07967E+01	8.81018E-01	1.52958E-03	0.00000E+00	0.00000E+00
271	8.33581E-01	1.08350E+01	8.80842E-01	1.53405E-03	0.00000E+00	0.00000E+00
272	8.52267E-01	1.08753E+01	8.80736E-01	1.53202E-03	0.00000E+00	0.00000E+00
273	9.12820E-01	1.09157E+01	8.80854E-01	1.53094E-03	0.00000E+00	0.00000E+00
274	8.86218E-01	1.09558E+01	8.80874E-01	1.52543E-03	0.00000E+00	0.00000E+00
275	8.65952E-01	1.09952E+01	8.80819E-01	1.52081E-03	0.00000E+00	0.00000E+00
276	8.83582E-01	1.10347E+01	8.80829E-01	1.51529E-03	0.00000E+00	0.00000E+00
277	8.95305E-01	1.10758E+01	8.80882E-01	1.51068E-03	0.00000E+00	0.00000E+00
278	8.58409E-01	1.11152E+01	8.80801E-01	1.50740E-03	0.00000E+00	0.00000E+00
279	9.31448E-01	1.11555E+01	8.80983E-01	1.51304E-03	0.00000E+00	0.00000E+00
280	8.90012E-01	1.11938E+01	8.81016E-01	1.50793E-03	0.00000E+00	0.00000E+00
281	8.83008E-01	1.12342E+01	8.81023E-01	1.50254E-03	0.00000E+00	0.00000E+00
282	8.77555E-01	1.12735E+01	8.81011E-01	1.49721E-03	0.00000E+00	0.00000E+00
283	8.64853E-01	1.13138E+01	8.80953E-01	1.49298E-03	0.00000E+00	0.00000E+00
284	8.86038E-01	1.13542E+01	8.80971E-01	1.48779E-03	0.00000E+00	0.00000E+00
285	8.78313E-01	1.13943E+01	8.80962E-01	1.48255E-03	0.00000E+00	0.00000E+00
286	9.09924E-01	1.14337E+01	8.81064E-01	1.48084E-03	0.00000E+00	0.00000E+00
287	8.69042E-01	1.14732E+01	8.81022E-01	1.47623E-03	0.00000E+00	0.00000E+00
288	8.76127E-01	1.15125E+01	8.81004E-01	1.47116E-03	0.00000E+00	0.00000E+00
289	8.82672E-01	1.15508E+01	8.81010E-01	1.46604E-03	0.00000E+00	0.00000E+00
290	8.55173E-01	1.15903E+01	8.80921E-01	1.46369E-03	0.00000E+00	0.00000E+00
291	9.01395E-01	1.16305E+01	8.80991E-01	1.46034E-03	0.00000E+00	0.00000E+00
292	8.88075E-01	1.16708E+01	8.81016E-01	1.45550E-03	0.00000E+00	0.00000E+00
293	8.86500E-01	1.17093E+01	8.81035E-01	1.45061E-03	0.00000E+00	0.00000E+00
294	8.80978E-01	1.17487E+01	8.81035E-01	1.44563E-03	0.00000E+00	0.00000E+00
295	9.10626E-01	1.17890E+01	8.81135E-01	1.44423E-03	0.00000E+00	0.00000E+00
296	8.52665E-01	1.18292E+01	8.81039E-01	1.44256E-03	0.00000E+00	0.00000E+00
297	8.91406E-01	1.18695E+01	8.81074E-01	1.43809E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

298	8.97473E-01	1.19098E+01	8.81129E-01	1.43430E-03	0.00000E+00	0.00000E+00
299	8.28838E-01	1.19500E+01	8.80953E-01	1.44026E-03	0.00000E+00	0.00000E+00
300	9.47763E-01	1.19903E+01	8.81177E-01	1.45282E-03	0.00000E+00	0.00000E+00
301	9.02023E-01	1.20315E+01	8.81247E-01	1.44963E-03	0.00000E+00	0.00000E+00
302	8.49101E-01	1.20718E+01	8.81140E-01	1.44876E-03	0.00000E+00	0.00000E+00
303	8.66354E-01	1.21112E+01	8.81091E-01	1.44477E-03	0.00000E+00	0.00000E+00
304	8.63843E-01	1.21505E+01	8.81034E-01	1.44111E-03	0.00000E+00	0.00000E+00
305	8.77762E-01	1.21898E+01	8.81023E-01	1.43639E-03	0.00000E+00	0.00000E+00
306	9.13651E-01	1.22292E+01	8.81130E-01	1.43568E-03	0.00000E+00	0.00000E+00
307	8.80096E-01	1.22687E+01	8.81127E-01	1.43096E-03	0.00000E+00	0.00000E+00
308	8.97984E-01	1.23088E+01	8.81182E-01	1.42734E-03	0.00000E+00	0.00000E+00
309	8.77844E-01	1.23482E+01	8.81171E-01	1.42273E-03	0.00000E+00	0.00000E+00
310	8.76024E-01	1.23877E+01	8.81154E-01	1.41820E-03	0.00000E+00	0.00000E+00
311	8.27738E-01	1.24270E+01	8.80981E-01	1.42413E-03	0.00000E+00	0.00000E+00
312	9.37744E-01	1.24673E+01	8.81165E-01	1.43129E-03	0.00000E+00	0.00000E+00
313	8.85321E-01	1.25057E+01	8.81178E-01	1.42675E-03	0.00000E+00	0.00000E+00
314	8.97061E-01	1.25460E+01	8.81229E-01	1.42308E-03	0.00000E+00	0.00000E+00
315	8.66558E-01	1.25845E+01	8.81182E-01	1.41930E-03	0.00000E+00	0.00000E+00
316	9.08470E-01	1.26247E+01	8.81269E-01	1.41744E-03	0.00000E+00	0.00000E+00
317	9.40257E-01	1.26640E+01	8.81456E-01	1.42528E-03	0.00000E+00	0.00000E+00
318	8.89599E-01	1.27035E+01	8.81482E-01	1.42100E-03	0.00000E+00	0.00000E+00
319	9.10267E-01	1.27428E+01	8.81573E-01	1.41942E-03	0.00000E+00	0.00000E+00
320	9.47962E-01	1.27830E+01	8.81781E-01	1.43027E-03	0.00000E+00	0.00000E+00
321	8.63157E-01	1.28225E+01	8.81723E-01	1.42697E-03	0.00000E+00	0.00000E+00
322	8.87308E-01	1.28608E+01	8.81741E-01	1.42261E-03	0.00000E+00	0.00000E+00
323	8.78837E-01	1.29012E+01	8.81731E-01	1.41820E-03	0.00000E+00	0.00000E+00
324	8.69994E-01	1.29405E+01	8.81695E-01	1.41426E-03	0.00000E+00	0.00000E+00
325	8.48419E-01	1.29808E+01	8.81592E-01	1.41363E-03	0.00000E+00	0.00000E+00
326	8.24851E-01	1.30212E+01	8.81417E-01	1.42010E-03	0.00000E+00	0.00000E+00
327	8.97920E-01	1.30605E+01	8.81468E-01	1.41664E-03	0.00000E+00	0.00000E+00
328	8.87696E-01	1.30998E+01	8.81487E-01	1.41241E-03	0.00000E+00	0.00000E+00
329	8.92020E-01	1.31410E+01	8.81519E-01	1.40846E-03	0.00000E+00	0.00000E+00
330	8.71113E-01	1.31785E+01	8.81487E-01	1.40452E-03	0.00000E+00	0.00000E+00
331	8.52141E-01	1.32188E+01	8.81398E-01	1.40308E-03	0.00000E+00	0.00000E+00
332	8.74587E-01	1.32592E+01	8.81377E-01	1.39897E-03	0.00000E+00	0.00000E+00
333	9.20351E-01	1.32993E+01	8.81495E-01	1.39970E-03	0.00000E+00	0.00000E+00
334	8.33038E-01	1.33387E+01	8.81349E-01	1.40309E-03	0.00000E+00	0.00000E+00
335	8.68112E-01	1.33790E+01	8.81309E-01	1.39943E-03	0.00000E+00	0.00000E+00
336	9.17371E-01	1.34175E+01	8.81417E-01	1.39941E-03	0.00000E+00	0.00000E+00
337	9.08943E-01	1.34587E+01	8.81500E-01	1.39764E-03	0.00000E+00	0.00000E+00
338	8.58098E-01	1.34990E+01	8.81430E-01	1.39522E-03	0.00000E+00	0.00000E+00
339	8.67702E-01	1.35383E+01	8.81389E-01	1.39167E-03	0.00000E+00	0.00000E+00
340	8.96813E-01	1.35785E+01	8.81435E-01	1.38829E-03	0.00000E+00	0.00000E+00
341	9.18136E-01	1.36180E+01	8.81543E-01	1.38842E-03	0.00000E+00	0.00000E+00
342	8.95133E-01	1.36582E+01	8.81583E-01	1.38491E-03	0.00000E+00	0.00000E+00
343	9.00867E-01	1.36985E+01	8.81640E-01	1.38200E-03	0.00000E+00	0.00000E+00
344	8.94506E-01	1.37388E+01	8.81677E-01	1.37846E-03	0.00000E+00	0.00000E+00
345	8.94107E-01	1.37772E+01	8.81713E-01	1.37492E-03	0.00000E+00	0.00000E+00
346	8.91497E-01	1.38157E+01	8.81742E-01	1.37121E-03	0.00000E+00	0.00000E+00
347	8.80193E-01	1.38550E+01	8.81737E-01	1.36724E-03	0.00000E+00	0.00000E+00
348	8.99451E-01	1.38943E+01	8.81789E-01	1.36424E-03	0.00000E+00	0.00000E+00
349	8.68768E-01	1.39347E+01	8.81751E-01	1.36082E-03	0.00000E+00	0.00000E+00
350	8.73630E-01	1.39758E+01	8.81728E-01	1.35710E-03	0.00000E+00	0.00000E+00
351	8.82763E-01	1.40162E+01	8.81731E-01	1.35321E-03	0.00000E+00	0.00000E+00
352	8.84028E-01	1.40555E+01	8.81737E-01	1.34936E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

353	9.03673E-01	1.40958E+01	8.81800E-01	1.34696E-03	0.00000E+00	0.00000E+00
354	8.78421E-01	1.41360E+01	8.81790E-01	1.34316E-03	0.00000E+00	0.00000E+00
355	8.66420E-01	1.41755E+01	8.81747E-01	1.34006E-03	0.00000E+00	0.00000E+00
356	8.70497E-01	1.42157E+01	8.81715E-01	1.33665E-03	0.00000E+00	0.00000E+00
357	9.06568E-01	1.42560E+01	8.81785E-01	1.33471E-03	0.00000E+00	0.00000E+00
358	8.84640E-01	1.42953E+01	8.81793E-01	1.33098E-03	0.00000E+00	0.00000E+00
359	8.86011E-01	1.43365E+01	8.81805E-01	1.32730E-03	0.00000E+00	0.00000E+00
360	8.44734E-01	1.43750E+01	8.81701E-01	1.32763E-03	0.00000E+00	0.00000E+00
361	8.90264E-01	1.44153E+01	8.81725E-01	1.32414E-03	0.00000E+00	0.00000E+00
362	8.87351E-01	1.44555E+01	8.81741E-01	1.32055E-03	0.00000E+00	0.00000E+00
363	8.98732E-01	1.44958E+01	8.81788E-01	1.31773E-03	0.00000E+00	0.00000E+00
364	8.56659E-01	1.45352E+01	8.81718E-01	1.31592E-03	0.00000E+00	0.00000E+00
365	8.96493E-01	1.45745E+01	8.81759E-01	1.31292E-03	0.00000E+00	0.00000E+00
366	9.01091E-01	1.46158E+01	8.81812E-01	1.31038E-03	0.00000E+00	0.00000E+00
367	8.77226E-01	1.46560E+01	8.81800E-01	1.30685E-03	0.00000E+00	0.00000E+00
368	8.87962E-01	1.46953E+01	8.81816E-01	1.30338E-03	0.00000E+00	0.00000E+00
369	9.05514E-01	1.47348E+01	8.81881E-01	1.30143E-03	0.00000E+00	0.00000E+00
370	8.70202E-01	1.47742E+01	8.81849E-01	1.29828E-03	0.00000E+00	0.00000E+00
371	8.99787E-01	1.48135E+01	8.81898E-01	1.29566E-03	0.00000E+00	0.00000E+00
372	8.72817E-01	1.48547E+01	8.81873E-01	1.29239E-03	0.00000E+00	0.00000E+00
373	8.79512E-01	1.48940E+01	8.81867E-01	1.28892E-03	0.00000E+00	0.00000E+00
374	9.08171E-01	1.49333E+01	8.81938E-01	1.28739E-03	0.00000E+00	0.00000E+00
375	9.09237E-01	1.49728E+01	8.82011E-01	1.28602E-03	0.00000E+00	0.00000E+00
376	9.28986E-01	1.50122E+01	8.82136E-01	1.28871E-03	0.00000E+00	0.00000E+00
377	8.86542E-01	1.50515E+01	8.82148E-01	1.28533E-03	0.00000E+00	0.00000E+00
378	8.65032E-01	1.50900E+01	8.82103E-01	1.28271E-03	0.00000E+00	0.00000E+00
379	8.58366E-01	1.51302E+01	8.82040E-01	1.28085E-03	0.00000E+00	0.00000E+00
380	8.71339E-01	1.51697E+01	8.82011E-01	1.27777E-03	0.00000E+00	0.00000E+00
381	9.18535E-01	1.52098E+01	8.82108E-01	1.27803E-03	0.00000E+00	0.00000E+00
382	8.61027E-01	1.52492E+01	8.82052E-01	1.27587E-03	0.00000E+00	0.00000E+00
383	8.84519E-01	1.52887E+01	8.82059E-01	1.27254E-03	0.00000E+00	0.00000E+00
384	8.77978E-01	1.53288E+01	8.82048E-01	1.26925E-03	0.00000E+00	0.00000E+00
385	8.98160E-01	1.53692E+01	8.82090E-01	1.26663E-03	0.00000E+00	0.00000E+00
386	8.63618E-01	1.54085E+01	8.82042E-01	1.26424E-03	0.00000E+00	0.00000E+00
387	9.01063E-01	1.54470E+01	8.82091E-01	1.26192E-03	0.00000E+00	0.00000E+00
388	8.90304E-01	1.54863E+01	8.82113E-01	1.25883E-03	0.00000E+00	0.00000E+00
389	8.91824E-01	1.55267E+01	8.82138E-01	1.25582E-03	0.00000E+00	0.00000E+00
390	8.46748E-01	1.55650E+01	8.82047E-01	1.25589E-03	0.00000E+00	0.00000E+00
391	8.88290E-01	1.56043E+01	8.82063E-01	1.25277E-03	0.00000E+00	0.00000E+00
392	8.88432E-01	1.56428E+01	8.82079E-01	1.24966E-03	0.00000E+00	0.00000E+00
393	8.67039E-01	1.56822E+01	8.82041E-01	1.24705E-03	0.00000E+00	0.00000E+00
394	8.87371E-01	1.57225E+01	8.82054E-01	1.24394E-03	0.00000E+00	0.00000E+00
395	9.11865E-01	1.57628E+01	8.82130E-01	1.24308E-03	0.00000E+00	0.00000E+00
396	8.71234E-01	1.58012E+01	8.82102E-01	1.24023E-03	0.00000E+00	0.00000E+00
397	9.15677E-01	1.58407E+01	8.82187E-01	1.24001E-03	0.00000E+00	0.00000E+00
398	8.89985E-01	1.58808E+01	8.82207E-01	1.23703E-03	0.00000E+00	0.00000E+00
399	8.90441E-01	1.59212E+01	8.82228E-01	1.23408E-03	0.00000E+00	0.00000E+00
400	8.75305E-01	1.59605E+01	8.82210E-01	1.23110E-03	0.00000E+00	0.00000E+00
401	8.87245E-01	1.59998E+01	8.82223E-01	1.22808E-03	0.00000E+00	0.00000E+00
402	9.05891E-01	1.60392E+01	8.82282E-01	1.22643E-03	0.00000E+00	0.00000E+00
403	8.70886E-01	1.60787E+01	8.82254E-01	1.22370E-03	0.00000E+00	0.00000E+00
404	8.79554E-01	1.61188E+01	8.82247E-01	1.22067E-03	0.00000E+00	0.00000E+00
405	8.39682E-01	1.61582E+01	8.82141E-01	1.22221E-03	0.00000E+00	0.00000E+00
406	9.03185E-01	1.61985E+01	8.82193E-01	1.22029E-03	0.00000E+00	0.00000E+00
407	8.56177E-01	1.62388E+01	8.82129E-01	1.21897E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

408	8.84160E-01	1.62782E+01	8.82134E-01	1.21597E-03	0.00000E+00	0.00000E+00
409	9.07301E-01	1.63185E+01	8.82196E-01	1.21456E-03	0.00000E+00	0.00000E+00
410	8.68245E-01	1.63568E+01	8.82162E-01	1.21206E-03	0.00000E+00	0.00000E+00
411	9.07611E-01	1.63963E+01	8.82224E-01	1.21069E-03	0.00000E+00	0.00000E+00
412	8.75487E-01	1.64365E+01	8.82208E-01	1.20785E-03	0.00000E+00	0.00000E+00
413	9.11941E-01	1.64758E+01	8.82280E-01	1.20707E-03	0.00000E+00	0.00000E+00
414	8.55267E-01	1.65153E+01	8.82214E-01	1.20592E-03	0.00000E+00	0.00000E+00
415	8.68168E-01	1.65555E+01	8.82180E-01	1.20348E-03	0.00000E+00	0.00000E+00
416	9.15813E-01	1.65958E+01	8.82262E-01	1.20332E-03	0.00000E+00	0.00000E+00
417	8.99200E-01	1.66343E+01	8.82302E-01	1.20111E-03	0.00000E+00	0.00000E+00
418	8.96912E-01	1.66745E+01	8.82338E-01	1.19873E-03	0.00000E+00	0.00000E+00
419	8.62986E-01	1.67138E+01	8.82291E-01	1.19675E-03	0.00000E+00	0.00000E+00
420	8.96614E-01	1.67523E+01	8.82325E-01	1.19438E-03	0.00000E+00	0.00000E+00
421	8.82352E-01	1.67927E+01	8.82326E-01	1.19152E-03	0.00000E+00	0.00000E+00
422	8.65613E-01	1.68328E+01	8.82286E-01	1.18935E-03	0.00000E+00	0.00000E+00
423	8.82556E-01	1.68723E+01	8.82286E-01	1.18652E-03	0.00000E+00	0.00000E+00
424	9.08713E-01	1.69117E+01	8.82349E-01	1.18536E-03	0.00000E+00	0.00000E+00
425	8.93695E-01	1.69502E+01	8.82376E-01	1.18286E-03	0.00000E+00	0.00000E+00
426	8.70718E-01	1.69895E+01	8.82348E-01	1.18039E-03	0.00000E+00	0.00000E+00
427	9.06052E-01	1.70288E+01	8.82404E-01	1.17893E-03	0.00000E+00	0.00000E+00
428	8.90895E-01	1.70682E+01	8.82424E-01	1.17633E-03	0.00000E+00	0.00000E+00
429	8.72919E-01	1.71085E+01	8.82402E-01	1.17378E-03	0.00000E+00	0.00000E+00
430	8.85284E-01	1.71468E+01	8.82408E-01	1.17105E-03	0.00000E+00	0.00000E+00
431	8.91349E-01	1.71872E+01	8.82429E-01	1.16850E-03	0.00000E+00	0.00000E+00
432	8.57679E-01	1.72275E+01	8.82372E-01	1.16720E-03	0.00000E+00	0.00000E+00
433	8.81645E-01	1.72677E+01	8.82370E-01	1.16449E-03	0.00000E+00	0.00000E+00
434	9.15060E-01	1.73072E+01	8.82446E-01	1.16426E-03	0.00000E+00	0.00000E+00
435	8.39969E-01	1.73473E+01	8.82348E-01	1.16570E-03	0.00000E+00	0.00000E+00
436	9.40917E-01	1.73867E+01	8.82483E-01	1.17082E-03	0.00000E+00	0.00000E+00
437	8.72358E-01	1.74270E+01	8.82459E-01	1.16835E-03	0.00000E+00	0.00000E+00
438	9.35555E-01	1.74663E+01	8.82581E-01	1.17201E-03	0.00000E+00	0.00000E+00
439	8.53128E-01	1.75058E+01	8.82514E-01	1.17127E-03	0.00000E+00	0.00000E+00
440	8.54364E-01	1.75452E+01	8.82449E-01	1.17036E-03	0.00000E+00	0.00000E+00
441	8.48884E-01	1.75845E+01	8.82373E-01	1.17019E-03	0.00000E+00	0.00000E+00
442	8.92398E-01	1.76248E+01	8.82396E-01	1.16775E-03	0.00000E+00	0.00000E+00
443	8.92706E-01	1.76660E+01	8.82419E-01	1.16533E-03	0.00000E+00	0.00000E+00
444	8.41588E-01	1.77062E+01	8.82327E-01	1.16636E-03	0.00000E+00	0.00000E+00
445	8.77350E-01	1.77465E+01	8.82316E-01	1.16378E-03	0.00000E+00	0.00000E+00
446	8.61286E-01	1.77858E+01	8.82268E-01	1.16212E-03	0.00000E+00	0.00000E+00
447	9.02126E-01	1.78252E+01	8.82313E-01	1.16036E-03	0.00000E+00	0.00000E+00
448	9.04592E-01	1.78655E+01	8.82363E-01	1.15883E-03	0.00000E+00	0.00000E+00
449	8.97089E-01	1.79058E+01	8.82396E-01	1.15671E-03	0.00000E+00	0.00000E+00
450	8.76106E-01	1.79452E+01	8.82382E-01	1.15421E-03	0.00000E+00	0.00000E+00
451	8.76715E-01	1.79837E+01	8.82369E-01	1.15170E-03	0.00000E+00	0.00000E+00
452	8.86842E-01	1.80230E+01	8.82379E-01	1.14918E-03	0.00000E+00	0.00000E+00
453	8.95730E-01	1.80632E+01	8.82409E-01	1.14702E-03	0.00000E+00	0.00000E+00
454	8.74973E-01	1.81027E+01	8.82392E-01	1.14459E-03	0.00000E+00	0.00000E+00
455	8.75774E-01	1.81420E+01	8.82378E-01	1.14216E-03	0.00000E+00	0.00000E+00
456	8.41602E-01	1.81813E+01	8.82288E-01	1.14317E-03	0.00000E+00	0.00000E+00
457	8.75021E-01	1.82217E+01	8.82272E-01	1.14077E-03	0.00000E+00	0.00000E+00
458	8.73079E-01	1.82610E+01	8.82252E-01	1.13844E-03	0.00000E+00	0.00000E+00
459	8.32778E-01	1.83003E+01	8.82143E-01	1.14110E-03	0.00000E+00	0.00000E+00
460	8.44397E-01	1.83397E+01	8.82061E-01	1.14158E-03	0.00000E+00	0.00000E+00
461	8.38710E-01	1.83790E+01	8.81966E-01	1.14300E-03	0.00000E+00	0.00000E+00
462	8.82149E-01	1.84185E+01	8.81967E-01	1.14051E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

463	9.06293E-01	1.84587E+01	8.82020E-01	1.13926E-03	0.00000E+00	0.00000E+00
464	8.49087E-01	1.84990E+01	8.81948E-01	1.13902E-03	0.00000E+00	0.00000E+00
465	8.55269E-01	1.85383E+01	8.81891E-01	1.13802E-03	0.00000E+00	0.00000E+00
466	8.89566E-01	1.85787E+01	8.81907E-01	1.13568E-03	0.00000E+00	0.00000E+00
467	8.92740E-01	1.86188E+01	8.81931E-01	1.13348E-03	0.00000E+00	0.00000E+00
468	8.39615E-01	1.86573E+01	8.81840E-01	1.13468E-03	0.00000E+00	0.00000E+00
469	8.74764E-01	1.86985E+01	8.81825E-01	1.13235E-03	0.00000E+00	0.00000E+00
470	8.90642E-01	1.87388E+01	8.81843E-01	1.13009E-03	0.00000E+00	0.00000E+00
471	8.90220E-01	1.87782E+01	8.81861E-01	1.12782E-03	0.00000E+00	0.00000E+00
472	9.61537E-01	1.88175E+01	8.82031E-01	1.13811E-03	0.00000E+00	0.00000E+00
473	9.01940E-01	1.88552E+01	8.82073E-01	1.13648E-03	0.00000E+00	0.00000E+00
474	8.80985E-01	1.88953E+01	8.82071E-01	1.13407E-03	0.00000E+00	0.00000E+00
475	9.04594E-01	1.89347E+01	8.82118E-01	1.13267E-03	0.00000E+00	0.00000E+00
476	8.48742E-01	1.89750E+01	8.82048E-01	1.13247E-03	0.00000E+00	0.00000E+00
477	8.77631E-01	1.90143E+01	8.82039E-01	1.13012E-03	0.00000E+00	0.00000E+00
478	8.69849E-01	1.90537E+01	8.82013E-01	1.12804E-03	0.00000E+00	0.00000E+00
479	8.69407E-01	1.90932E+01	8.81987E-01	1.12598E-03	0.00000E+00	0.00000E+00
480	8.99173E-01	1.91325E+01	8.82023E-01	1.12420E-03	0.00000E+00	0.00000E+00
481	8.52328E-01	1.91727E+01	8.81961E-01	1.12356E-03	0.00000E+00	0.00000E+00
482	8.58089E-01	1.92122E+01	8.81911E-01	1.12232E-03	0.00000E+00	0.00000E+00
483	9.19698E-01	1.92515E+01	8.81989E-01	1.12273E-03	0.00000E+00	0.00000E+00
484	8.83794E-01	1.92917E+01	8.81993E-01	1.12041E-03	0.00000E+00	0.00000E+00
485	8.69506E-01	1.93312E+01	8.81967E-01	1.11838E-03	0.00000E+00	0.00000E+00
486	8.86563E-01	1.93713E+01	8.81977E-01	1.11611E-03	0.00000E+00	0.00000E+00
487	8.56677E-01	1.94098E+01	8.81925E-01	1.11503E-03	0.00000E+00	0.00000E+00
488	8.63728E-01	1.94502E+01	8.81887E-01	1.11336E-03	0.00000E+00	0.00000E+00
489	8.72829E-01	1.94903E+01	8.81869E-01	1.11123E-03	0.00000E+00	0.00000E+00
490	9.02072E-01	1.95288E+01	8.81910E-01	1.10972E-03	0.00000E+00	0.00000E+00
491	8.97118E-01	1.95692E+01	8.81941E-01	1.10789E-03	0.00000E+00	0.00000E+00
492	9.25546E-01	1.96093E+01	8.82030E-01	1.10920E-03	0.00000E+00	0.00000E+00
493	9.15487E-01	1.96488E+01	8.82098E-01	1.10903E-03	0.00000E+00	0.00000E+00
494	8.53320E-01	1.96890E+01	8.82040E-01	1.10832E-03	0.00000E+00	0.00000E+00
495	8.97121E-01	1.97275E+01	8.82070E-01	1.10649E-03	0.00000E+00	0.00000E+00
496	8.71942E-01	1.97678E+01	8.82050E-01	1.10444E-03	0.00000E+00	0.00000E+00
497	8.55946E-01	1.98080E+01	8.81997E-01	1.10347E-03	0.00000E+00	0.00000E+00
498	9.02029E-01	1.98473E+01	8.82038E-01	1.10198E-03	0.00000E+00	0.00000E+00
499	8.60758E-01	1.98868E+01	8.81995E-01	1.10060E-03	0.00000E+00	0.00000E+00
500	9.03307E-01	1.99270E+01	8.82037E-01	1.09922E-03	0.00000E+00	0.00000E+00
501	8.68025E-01	1.99663E+01	8.82009E-01	1.09737E-03	0.00000E+00	0.00000E+00
502	8.70763E-01	2.00067E+01	8.81987E-01	1.09541E-03	0.00000E+00	0.00000E+00
503	8.56129E-01	2.00460E+01	8.81935E-01	1.09444E-03	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132 WARNING... ONLY 985 INDEPENDENT FISSION POINTS WERE GENERATED						
504	8.17122E-01	2.00863E+01	8.81806E-01	1.09986E-03	0.00000E+00	0.00000E+00
505	8.70737E-01	2.01248E+01	8.81784E-01	1.09789E-03	0.00000E+00	0.00000E+00
506	9.09339E-01	2.01650E+01	8.81839E-01	1.09707E-03	0.00000E+00	0.00000E+00
507	9.19328E-01	2.02045E+01	8.81913E-01	1.09741E-03	0.00000E+00	0.00000E+00
508	9.12408E-01	2.02438E+01	8.81973E-01	1.09690E-03	0.00000E+00	0.00000E+00
509	8.78124E-01	2.02832E+01	8.81966E-01	1.09476E-03	0.00000E+00	0.00000E+00
510	8.79549E-01	2.03225E+01	8.81961E-01	1.09261E-03	0.00000E+00	0.00000E+00
511	9.31643E-01	2.03618E+01	8.82059E-01	1.09482E-03	0.00000E+00	0.00000E+00
512	8.86865E-01	2.04003E+01	8.82068E-01	1.09271E-03	0.00000E+00	0.00000E+00
513	8.84233E-01	2.04397E+01	8.82072E-01	1.09058E-03	0.00000E+00	0.00000E+00
514	9.00211E-01	2.04790E+01	8.82108E-01	1.08903E-03	0.00000E+00	0.00000E+00
515	9.21342E-01	2.05193E+01	8.82184E-01	1.08959E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

516	9.23000E-01	2.05587E+01	8.82264E-01	1.09036E-03	0.00000E+00	0.00000E+00
517	8.93883E-01	2.05990E+01	8.82286E-01	1.08848E-03	0.00000E+00	0.00000E+00
518	8.63400E-01	2.06393E+01	8.82250E-01	1.08698E-03	0.00000E+00	0.00000E+00
519	8.61607E-01	2.06795E+01	8.82210E-01	1.08561E-03	0.00000E+00	0.00000E+00
520	8.63854E-01	2.07188E+01	8.82174E-01	1.08409E-03	0.00000E+00	0.00000E+00
521	8.91655E-01	2.07592E+01	8.82192E-01	1.08216E-03	0.00000E+00	0.00000E+00
522	8.73559E-01	2.07977E+01	8.82176E-01	1.08020E-03	0.00000E+00	0.00000E+00
523	8.68968E-01	2.08360E+01	8.82151E-01	1.07842E-03	0.00000E+00	0.00000E+00
524	8.66051E-01	2.08745E+01	8.82120E-01	1.07680E-03	0.00000E+00	0.00000E+00
525	8.78692E-01	2.09157E+01	8.82113E-01	1.07476E-03	0.00000E+00	0.00000E+00
526	8.47636E-01	2.09568E+01	8.82047E-01	1.07472E-03	0.00000E+00	0.00000E+00
527	8.58411E-01	2.09963E+01	8.82002E-01	1.07361E-03	0.00000E+00	0.00000E+00
528	8.12478E-01	2.10365E+01	8.81870E-01	1.07969E-03	0.00000E+00	0.00000E+00
529	8.82227E-01	2.10768E+01	8.81871E-01	1.07764E-03	0.00000E+00	0.00000E+00
530	8.42374E-01	2.11172E+01	8.81796E-01	1.07820E-03	0.00000E+00	0.00000E+00
531	8.74614E-01	2.11573E+01	8.81782E-01	1.07624E-03	0.00000E+00	0.00000E+00
532	8.86053E-01	2.11967E+01	8.81790E-01	1.07424E-03	0.00000E+00	0.00000E+00
533	9.12332E-01	2.12370E+01	8.81848E-01	1.07376E-03	0.00000E+00	0.00000E+00
534	8.78468E-01	2.12763E+01	8.81842E-01	1.07176E-03	0.00000E+00	0.00000E+00
535	8.90432E-01	2.13167E+01	8.81858E-01	1.06986E-03	0.00000E+00	0.00000E+00
536	8.44220E-01	2.13570E+01	8.81787E-01	1.07018E-03	0.00000E+00	0.00000E+00
537	8.44108E-01	2.13963E+01	8.81717E-01	1.07050E-03	0.00000E+00	0.00000E+00
538	9.23963E-01	2.14357E+01	8.81796E-01	1.07140E-03	0.00000E+00	0.00000E+00
539	8.74314E-01	2.14750E+01	8.81782E-01	1.06950E-03	0.00000E+00	0.00000E+00
540	8.80722E-01	2.15143E+01	8.81780E-01	1.06751E-03	0.00000E+00	0.00000E+00
541	8.89401E-01	2.15538E+01	8.81794E-01	1.06562E-03	0.00000E+00	0.00000E+00
542	8.94953E-01	2.15932E+01	8.81818E-01	1.06392E-03	0.00000E+00	0.00000E+00
543	8.86277E-01	2.16333E+01	8.81827E-01	1.06199E-03	0.00000E+00	0.00000E+00
544	8.90504E-01	2.16737E+01	8.81843E-01	1.06015E-03	0.00000E+00	0.00000E+00
545	8.88598E-01	2.17122E+01	8.81855E-01	1.05827E-03	0.00000E+00	0.00000E+00
546	8.91285E-01	2.17505E+01	8.81872E-01	1.05646E-03	0.00000E+00	0.00000E+00
547	8.96664E-01	2.17908E+01	8.81899E-01	1.05487E-03	0.00000E+00	0.00000E+00
548	9.40468E-01	2.18293E+01	8.82007E-01	1.05839E-03	0.00000E+00	0.00000E+00
549	9.14963E-01	2.18705E+01	8.82067E-01	1.05817E-03	0.00000E+00	0.00000E+00
550	8.68124E-01	2.19090E+01	8.82042E-01	1.05654E-03	0.00000E+00	0.00000E+00
551	8.64745E-01	2.19483E+01	8.82010E-01	1.05508E-03	0.00000E+00	0.00000E+00
552	8.67584E-01	2.19877E+01	8.81984E-01	1.05349E-03	0.00000E+00	0.00000E+00
553	9.01171E-01	2.20288E+01	8.82019E-01	1.05215E-03	0.00000E+00	0.00000E+00
554	9.04750E-01	2.20700E+01	8.82060E-01	1.05105E-03	0.00000E+00	0.00000E+00
555	8.59889E-01	2.21095E+01	8.82020E-01	1.04992E-03	0.00000E+00	0.00000E+00
556	8.73067E-01	2.21478E+01	8.82004E-01	1.04814E-03	0.00000E+00	0.00000E+00
557	8.63804E-01	2.21863E+01	8.81971E-01	1.04677E-03	0.00000E+00	0.00000E+00
558	9.48995E-01	2.22257E+01	8.82091E-01	1.05181E-03	0.00000E+00	0.00000E+00
559	9.20640E-01	2.22660E+01	8.82160E-01	1.05220E-03	0.00000E+00	0.00000E+00
560	8.43694E-01	2.23062E+01	8.82092E-01	1.05258E-03	0.00000E+00	0.00000E+00
561	8.82129E-01	2.23457E+01	8.82092E-01	1.05069E-03	0.00000E+00	0.00000E+00
562	8.97580E-01	2.23850E+01	8.82119E-01	1.04918E-03	0.00000E+00	0.00000E+00
563	8.61908E-01	2.24243E+01	8.82083E-01	1.04792E-03	0.00000E+00	0.00000E+00
564	8.68591E-01	2.24628E+01	8.82059E-01	1.04633E-03	0.00000E+00	0.00000E+00
565	8.66369E-01	2.25030E+01	8.82031E-01	1.04485E-03	0.00000E+00	0.00000E+00
566	9.04905E-01	2.25425E+01	8.82072E-01	1.04378E-03	0.00000E+00	0.00000E+00
567	9.09679E-01	2.25827E+01	8.82121E-01	1.04308E-03	0.00000E+00	0.00000E+00
568	8.82234E-01	2.26230E+01	8.82121E-01	1.04123E-03	0.00000E+00	0.00000E+00
569	8.68176E-01	2.26623E+01	8.82096E-01	1.03968E-03	0.00000E+00	0.00000E+00
570	8.99375E-01	2.27027E+01	8.82127E-01	1.03830E-03	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

571	8.77770E-01	2.27420E+01	8.82119E-01	1.03650E-03	0.00000E+00	0.00000E+00
572	8.44679E-01	2.27832E+01	8.82053E-01	1.03676E-03	0.00000E+00	0.00000E+00
573	8.94850E-01	2.28225E+01	8.82076E-01	1.03519E-03	0.00000E+00	0.00000E+00
574	8.92159E-01	2.28628E+01	8.82094E-01	1.03353E-03	0.00000E+00	0.00000E+00
575	8.96787E-01	2.29032E+01	8.82119E-01	1.03204E-03	0.00000E+00	0.00000E+00
576	9.02736E-01	2.29443E+01	8.82155E-01	1.03087E-03	0.00000E+00	0.00000E+00
577	8.89806E-01	2.29845E+01	8.82168E-01	1.02916E-03	0.00000E+00	0.00000E+00
578	8.72142E-01	2.30240E+01	8.82151E-01	1.02752E-03	0.00000E+00	0.00000E+00
579	8.98391E-01	2.30642E+01	8.82179E-01	1.02612E-03	0.00000E+00	0.00000E+00
580	9.10201E-01	2.31027E+01	8.82228E-01	1.02549E-03	0.00000E+00	0.00000E+00
581	8.87118E-01	2.31420E+01	8.82236E-01	1.02375E-03	0.00000E+00	0.00000E+00
582	8.88775E-01	2.31823E+01	8.82247E-01	1.02205E-03	0.00000E+00	0.00000E+00
583	8.68045E-01	2.32217E+01	8.82223E-01	1.02058E-03	0.00000E+00	0.00000E+00
584	8.66961E-01	2.32620E+01	8.82197E-01	1.01916E-03	0.00000E+00	0.00000E+00
585	8.78780E-01	2.33022E+01	8.82191E-01	1.01743E-03	0.00000E+00	0.00000E+00
586	8.89640E-01	2.33425E+01	8.82204E-01	1.01577E-03	0.00000E+00	0.00000E+00
587	9.04818E-01	2.33810E+01	8.82242E-01	1.01477E-03	0.00000E+00	0.00000E+00
588	8.78231E-01	2.34212E+01	8.82235E-01	1.01306E-03	0.00000E+00	0.00000E+00
589	9.07292E-01	2.34615E+01	8.82278E-01	1.01223E-03	0.00000E+00	0.00000E+00
590	8.71047E-01	2.35000E+01	8.82259E-01	1.01069E-03	0.00000E+00	0.00000E+00
591	8.63600E-01	2.35393E+01	8.82227E-01	1.00947E-03	0.00000E+00	0.00000E+00
592	9.11831E-01	2.35787E+01	8.82277E-01	1.00900E-03	0.00000E+00	0.00000E+00
593	8.96332E-01	2.36190E+01	8.82301E-01	1.00757E-03	0.00000E+00	0.00000E+00
594	9.04535E-01	2.36592E+01	8.82339E-01	1.00657E-03	0.00000E+00	0.00000E+00
595	8.22890E-01	2.36995E+01	8.82239E-01	1.00986E-03	0.00000E+00	0.00000E+00
596	8.87285E-01	2.37388E+01	8.82247E-01	1.00819E-03	0.00000E+00	0.00000E+00
597	8.95198E-01	2.37800E+01	8.82269E-01	1.00673E-03	0.00000E+00	0.00000E+00
598	8.72589E-01	2.38195E+01	8.82253E-01	1.00518E-03	0.00000E+00	0.00000E+00
599	8.68900E-01	2.38597E+01	8.82230E-01	1.00374E-03	0.00000E+00	0.00000E+00
600	8.79766E-01	2.38990E+01	8.82226E-01	1.00207E-03	0.00000E+00	0.00000E+00
601	8.60220E-01	2.39393E+01	8.82189E-01	1.00107E-03	0.00000E+00	0.00000E+00
602	8.94022E-01	2.39805E+01	8.82209E-01	9.99592E-04	0.00000E+00	0.00000E+00
603	8.79061E-01	2.40235E+01	8.82204E-01	9.97942E-04	0.00000E+00	0.00000E+00
604	8.39274E-01	2.40675E+01	8.82132E-01	9.98831E-04	0.00000E+00	0.00000E+00
605	8.76488E-01	2.41097E+01	8.82123E-01	9.97218E-04	0.00000E+00	0.00000E+00
606	8.72621E-01	2.41517E+01	8.82107E-01	9.95690E-04	0.00000E+00	0.00000E+00
607	9.03733E-01	2.41967E+01	8.82143E-01	9.94685E-04	0.00000E+00	0.00000E+00
608	8.49784E-01	2.42397E+01	8.82090E-01	9.94477E-04	0.00000E+00	0.00000E+00
609	9.01798E-01	2.42817E+01	8.82122E-01	9.93368E-04	0.00000E+00	0.00000E+00
610	8.58867E-01	2.43248E+01	8.82084E-01	9.92470E-04	0.00000E+00	0.00000E+00
611	8.66619E-01	2.43668E+01	8.82059E-01	9.91164E-04	0.00000E+00	0.00000E+00
612	8.88378E-01	2.44108E+01	8.82069E-01	9.89592E-04	0.00000E+00	0.00000E+00
613	8.86050E-01	2.44547E+01	8.82075E-01	9.87993E-04	0.00000E+00	0.00000E+00
614	8.60997E-01	2.44968E+01	8.82041E-01	9.86978E-04	0.00000E+00	0.00000E+00
615	9.00311E-01	2.45398E+01	8.82071E-01	9.85818E-04	0.00000E+00	0.00000E+00
616	8.54085E-01	2.45838E+01	8.82025E-01	9.85266E-04	0.00000E+00	0.00000E+00
617	8.91856E-01	2.46260E+01	8.82041E-01	9.83792E-04	0.00000E+00	0.00000E+00
618	9.11750E-01	2.46690E+01	8.82089E-01	9.83377E-04	0.00000E+00	0.00000E+00
619	8.74224E-01	2.47128E+01	8.82077E-01	9.81865E-04	0.00000E+00	0.00000E+00
620	9.02907E-01	2.47550E+01	8.82110E-01	9.80854E-04	0.00000E+00	0.00000E+00
621	8.70416E-01	2.47980E+01	8.82092E-01	9.79450E-04	0.00000E+00	0.00000E+00
622	8.48224E-01	2.48402E+01	8.82037E-01	9.79394E-04	0.00000E+00	0.00000E+00
623	8.75961E-01	2.48822E+01	8.82027E-01	9.77864E-04	0.00000E+00	0.00000E+00
624	8.74830E-01	2.49272E+01	8.82016E-01	9.76360E-04	0.00000E+00	0.00000E+00
625	8.47952E-01	2.49702E+01	8.81961E-01	9.76323E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

626	8.81179E-01	2.50122E+01	8.81960E-01	9.74758E-04	0.00000E+00	0.00000E+00
627	8.96724E-01	2.50562E+01	8.81983E-01	9.73484E-04	0.00000E+00	0.00000E+00
628	8.79372E-01	2.50992E+01	8.81979E-01	9.71937E-04	0.00000E+00	0.00000E+00
629	8.69244E-01	2.51432E+01	8.81959E-01	9.70598E-04	0.00000E+00	0.00000E+00
630	8.40868E-01	2.51853E+01	8.81893E-01	9.71258E-04	0.00000E+00	0.00000E+00
631	8.80322E-01	2.52273E+01	8.81891E-01	9.69715E-04	0.00000E+00	0.00000E+00
632	9.07667E-01	2.52713E+01	8.81932E-01	9.69039E-04	0.00000E+00	0.00000E+00
633	8.89959E-01	2.53152E+01	8.81944E-01	9.67586E-04	0.00000E+00	0.00000E+00
634	8.65669E-01	2.53583E+01	8.81919E-01	9.66397E-04	0.00000E+00	0.00000E+00
635	8.91702E-01	2.54022E+01	8.81934E-01	9.64993E-04	0.00000E+00	0.00000E+00
636	8.66782E-01	2.54452E+01	8.81910E-01	9.63766E-04	0.00000E+00	0.00000E+00
637	8.88486E-01	2.54892E+01	8.81921E-01	9.62303E-04	0.00000E+00	0.00000E+00
638	8.74290E-01	2.55340E+01	8.81909E-01	9.60863E-04	0.00000E+00	0.00000E+00
639	9.21651E-01	2.55780E+01	8.81971E-01	9.61380E-04	0.00000E+00	0.00000E+00
640	8.92408E-01	2.56365E+01	8.81987E-01	9.60012E-04	0.00000E+00	0.00000E+00
641	8.63697E-01	2.56988E+01	8.81959E-01	9.58935E-04	0.00000E+00	0.00000E+00
642	8.38296E-01	2.57455E+01	8.81891E-01	9.59863E-04	0.00000E+00	0.00000E+00
643	8.90787E-01	2.57903E+01	8.81904E-01	9.58465E-04	0.00000E+00	0.00000E+00
644	8.83693E-01	2.58362E+01	8.81907E-01	9.56975E-04	0.00000E+00	0.00000E+00
645	8.55107E-01	2.58800E+01	8.81865E-01	9.56394E-04	0.00000E+00	0.00000E+00
646	8.74671E-01	2.59222E+01	8.81854E-01	9.54973E-04	0.00000E+00	0.00000E+00
647	9.14515E-01	2.59680E+01	8.81905E-01	9.54835E-04	0.00000E+00	0.00000E+00
648	8.87086E-01	2.60118E+01	8.81913E-01	9.53390E-04	0.00000E+00	0.00000E+00
649	9.12143E-01	2.60577E+01	8.81960E-01	9.53061E-04	0.00000E+00	0.00000E+00
650	8.67178E-01	2.61043E+01	8.81937E-01	9.51863E-04	0.00000E+00	0.00000E+00
651	8.59865E-01	2.61492E+01	8.81903E-01	9.51003E-04	0.00000E+00	0.00000E+00
652	8.89124E-01	2.61950E+01	8.81914E-01	9.49604E-04	0.00000E+00	0.00000E+00
653	8.68977E-01	2.62417E+01	8.81894E-01	9.48352E-04	0.00000E+00	0.00000E+00
654	8.87446E-01	2.62902E+01	8.81903E-01	9.46935E-04	0.00000E+00	0.00000E+00
655	8.78722E-01	2.63360E+01	8.81898E-01	9.45496E-04	0.00000E+00	0.00000E+00
656	8.54416E-01	2.63817E+01	8.81856E-01	9.44984E-04	0.00000E+00	0.00000E+00
657	8.85734E-01	2.64283E+01	8.81862E-01	9.43559E-04	0.00000E+00	0.00000E+00
658	8.89068E-01	2.64760E+01	8.81873E-01	9.42184E-04	0.00000E+00	0.00000E+00
659	8.54308E-01	2.65208E+01	8.81831E-01	9.41684E-04	0.00000E+00	0.00000E+00
660	8.96234E-01	2.65685E+01	8.81853E-01	9.40506E-04	0.00000E+00	0.00000E+00
661	8.82145E-01	2.66133E+01	8.81853E-01	9.39078E-04	0.00000E+00	0.00000E+00
662	8.63435E-01	2.66573E+01	8.81825E-01	9.38069E-04	0.00000E+00	0.00000E+00
663	8.60742E-01	2.67012E+01	8.81793E-01	9.37192E-04	0.00000E+00	0.00000E+00
664	8.80474E-01	2.67460E+01	8.81791E-01	9.35777E-04	0.00000E+00	0.00000E+00
665	9.24087E-01	2.67890E+01	8.81855E-01	9.36540E-04	0.00000E+00	0.00000E+00
666	8.89245E-01	2.68293E+01	8.81866E-01	9.35195E-04	0.00000E+00	0.00000E+00
667	9.02053E-01	2.68697E+01	8.81897E-01	9.34281E-04	0.00000E+00	0.00000E+00
668	9.07235E-01	2.69098E+01	8.81935E-01	9.33652E-04	0.00000E+00	0.00000E+00
669	9.15363E-01	2.69502E+01	8.81985E-01	9.33598E-04	0.00000E+00	0.00000E+00
670	8.80778E-01	2.69905E+01	8.81983E-01	9.32201E-04	0.00000E+00	0.00000E+00
671	9.00666E-01	2.70307E+01	8.82011E-01	9.31225E-04	0.00000E+00	0.00000E+00
672	9.02518E-01	2.70710E+01	8.82041E-01	9.30338E-04	0.00000E+00	0.00000E+00
673	8.60191E-01	2.71103E+01	8.82009E-01	9.29521E-04	0.00000E+00	0.00000E+00
674	8.43162E-01	2.71498E+01	8.81951E-01	9.29935E-04	0.00000E+00	0.00000E+00
675	8.54494E-01	2.71892E+01	8.81910E-01	9.29448E-04	0.00000E+00	0.00000E+00
676	9.00082E-01	2.72285E+01	8.81937E-01	9.28460E-04	0.00000E+00	0.00000E+00
677	8.76369E-01	2.72670E+01	8.81929E-01	9.27120E-04	0.00000E+00	0.00000E+00
678	9.21882E-01	2.73072E+01	8.81988E-01	9.27632E-04	0.00000E+00	0.00000E+00
679	8.66146E-01	2.73475E+01	8.81965E-01	9.26556E-04	0.00000E+00	0.00000E+00
680	9.41145E-01	2.73878E+01	8.82052E-01	9.29297E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

681	9.21572E-01	2.74272E+01	8.82110E-01	9.29751E-04	0.00000E+00	0.00000E+00
682	8.82706E-01	2.74673E+01	8.82111E-01	9.28383E-04	0.00000E+00	0.00000E+00
683	9.30870E-01	2.75077E+01	8.82183E-01	9.29780E-04	0.00000E+00	0.00000E+00
684	9.18676E-01	2.75818E+01	8.82236E-01	9.29956E-04	0.00000E+00	0.00000E+00
685	8.72945E-01	2.76222E+01	8.82223E-01	9.28693E-04	0.00000E+00	0.00000E+00
686	8.96501E-01	2.76615E+01	8.82243E-01	9.27570E-04	0.00000E+00	0.00000E+00
687	8.84210E-01	2.77960E+01	8.82246E-01	9.26219E-04	0.00000E+00	0.00000E+00
688	8.55255E-01	2.79398E+01	8.82207E-01	9.25704E-04	0.00000E+00	0.00000E+00
689	9.27962E-01	2.79792E+01	8.82274E-01	9.26752E-04	0.00000E+00	0.00000E+00
690	8.87273E-01	2.81960E+01	8.82281E-01	9.25433E-04	0.00000E+00	0.00000E+00
691	8.73215E-01	2.82757E+01	8.82268E-01	9.24182E-04	0.00000E+00	0.00000E+00
692	8.60825E-01	2.83160E+01	8.82237E-01	9.23365E-04	0.00000E+00	0.00000E+00
693	8.83965E-01	2.83710E+01	8.82239E-01	9.22031E-04	0.00000E+00	0.00000E+00
694	8.60877E-01	2.85503E+01	8.82208E-01	9.21215E-04	0.00000E+00	0.00000E+00
695	8.61498E-01	2.87912E+01	8.82178E-01	9.20370E-04	0.00000E+00	0.00000E+00
696	8.61356E-01	2.89998E+01	8.82148E-01	9.19533E-04	0.00000E+00	0.00000E+00
697	8.77909E-01	2.91883E+01	8.82142E-01	9.18229E-04	0.00000E+00	0.00000E+00
698	9.06079E-01	2.93550E+01	8.82177E-01	9.17553E-04	0.00000E+00	0.00000E+00
699	8.99207E-01	2.94987E+01	8.82201E-01	9.16562E-04	0.00000E+00	0.00000E+00
700	9.05328E-01	2.95382E+01	8.82234E-01	9.15847E-04	0.00000E+00	0.00000E+00
701	8.97019E-01	2.95775E+01	8.82255E-01	9.14781E-04	0.00000E+00	0.00000E+00
702	8.86855E-01	2.96168E+01	8.82262E-01	9.13497E-04	0.00000E+00	0.00000E+00
703	8.88162E-01	2.96553E+01	8.82270E-01	9.12231E-04	0.00000E+00	0.00000E+00
704	8.78811E-01	2.96955E+01	8.82265E-01	9.10944E-04	0.00000E+00	0.00000E+00
705	9.02841E-01	2.97358E+01	8.82295E-01	9.10118E-04	0.00000E+00	0.00000E+00
706	8.86654E-01	2.97762E+01	8.82301E-01	9.08846E-04	0.00000E+00	0.00000E+00
707	8.27798E-01	2.98163E+01	8.82224E-01	9.10842E-04	0.00000E+00	0.00000E+00
708	9.01544E-01	2.98567E+01	8.82251E-01	9.09963E-04	0.00000E+00	0.00000E+00
709	8.88350E-01	2.98970E+01	8.82260E-01	9.08716E-04	0.00000E+00	0.00000E+00
710	8.47905E-01	2.99363E+01	8.82211E-01	9.08728E-04	0.00000E+00	0.00000E+00
711	8.85558E-01	2.99757E+01	8.82216E-01	9.07458E-04	0.00000E+00	0.00000E+00
712	8.65596E-01	3.00168E+01	8.82192E-01	9.06481E-04	0.00000E+00	0.00000E+00
713	8.86826E-01	3.00543E+01	8.82199E-01	9.05229E-04	0.00000E+00	0.00000E+00
714	8.84887E-01	3.00947E+01	8.82203E-01	9.03964E-04	0.00000E+00	0.00000E+00
715	8.66677E-01	3.01340E+01	8.82181E-01	9.02958E-04	0.00000E+00	0.00000E+00
716	9.05605E-01	3.01733E+01	8.82214E-01	9.02289E-04	0.00000E+00	0.00000E+00
717	8.80338E-01	3.02147E+01	8.82211E-01	9.01030E-04	0.00000E+00	0.00000E+00
718	9.11642E-01	3.02540E+01	8.82252E-01	9.00709E-04	0.00000E+00	0.00000E+00
719	8.58942E-01	3.02942E+01	8.82220E-01	9.00039E-04	0.00000E+00	0.00000E+00
720	8.78557E-01	3.03337E+01	8.82214E-01	8.98799E-04	0.00000E+00	0.00000E+00
721	8.92727E-01	3.03730E+01	8.82229E-01	8.97668E-04	0.00000E+00	0.00000E+00
722	8.84814E-01	3.04113E+01	8.82233E-01	8.96427E-04	0.00000E+00	0.00000E+00
723	8.93520E-01	3.04517E+01	8.82248E-01	8.95320E-04	0.00000E+00	0.00000E+00
724	8.78659E-01	3.04910E+01	8.82243E-01	8.94093E-04	0.00000E+00	0.00000E+00
725	8.61449E-01	3.05295E+01	8.82215E-01	8.93318E-04	0.00000E+00	0.00000E+00
726	8.99401E-01	3.05688E+01	8.82238E-01	8.92400E-04	0.00000E+00	0.00000E+00
727	8.80250E-01	3.06100E+01	8.82236E-01	8.91172E-04	0.00000E+00	0.00000E+00
728	9.15883E-01	3.06495E+01	8.82282E-01	8.91150E-04	0.00000E+00	0.00000E+00
729	8.86874E-01	3.06897E+01	8.82288E-01	8.89945E-04	0.00000E+00	0.00000E+00
730	8.63398E-01	3.07290E+01	8.82262E-01	8.89101E-04	0.00000E+00	0.00000E+00
731	8.69945E-01	3.07703E+01	8.82245E-01	8.88041E-04	0.00000E+00	0.00000E+00
732	8.43399E-01	3.08087E+01	8.82192E-01	8.88419E-04	0.00000E+00	0.00000E+00
733	8.80263E-01	3.08490E+01	8.82190E-01	8.87207E-04	0.00000E+00	0.00000E+00
734	9.32397E-01	3.08883E+01	8.82258E-01	8.88645E-04	0.00000E+00	0.00000E+00
735	8.37599E-01	3.09287E+01	8.82197E-01	8.89521E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

736	9.27630E-01	3.09680E+01	8.82259E-01	8.90462E-04	0.00000E+00	0.00000E+00
737	8.93560E-01	3.10083E+01	8.82275E-01	8.89382E-04	0.00000E+00	0.00000E+00
738	8.74518E-01	3.10477E+01	8.82264E-01	8.88236E-04	0.00000E+00	0.00000E+00
739	8.89837E-01	3.10870E+01	8.82274E-01	8.87089E-04	0.00000E+00	0.00000E+00
740	8.95250E-01	3.11263E+01	8.82292E-01	8.86061E-04	0.00000E+00	0.00000E+00
741	8.90671E-01	3.11657E+01	8.82303E-01	8.84934E-04	0.00000E+00	0.00000E+00
742	9.08882E-01	3.12042E+01	8.82339E-01	8.84467E-04	0.00000E+00	0.00000E+00
743	8.81208E-01	3.12435E+01	8.82338E-01	8.83273E-04	0.00000E+00	0.00000E+00
744	8.66451E-01	3.12838E+01	8.82316E-01	8.82342E-04	0.00000E+00	0.00000E+00
745	8.88192E-01	3.13232E+01	8.82324E-01	8.81189E-04	0.00000E+00	0.00000E+00
746	9.00612E-01	3.13617E+01	8.82349E-01	8.80347E-04	0.00000E+00	0.00000E+00
747	8.93411E-01	3.14000E+01	8.82364E-01	8.79290E-04	0.00000E+00	0.00000E+00
748	8.71527E-01	3.14395E+01	8.82349E-01	8.78231E-04	0.00000E+00	0.00000E+00
749	8.73503E-01	3.14788E+01	8.82337E-01	8.77134E-04	0.00000E+00	0.00000E+00
750	9.01890E-01	3.15182E+01	8.82363E-01	8.76351E-04	0.00000E+00	0.00000E+00
751	9.04663E-01	3.15575E+01	8.82393E-01	8.75686E-04	0.00000E+00	0.00000E+00
752	8.95468E-01	3.15968E+01	8.82410E-01	8.74692E-04	0.00000E+00	0.00000E+00
753	8.93231E-01	3.16363E+01	8.82425E-01	8.73645E-04	0.00000E+00	0.00000E+00
754	9.01668E-01	3.16765E+01	8.82450E-01	8.72858E-04	0.00000E+00	0.00000E+00
755	8.86805E-01	3.17177E+01	8.82456E-01	8.71717E-04	0.00000E+00	0.00000E+00
756	8.72417E-01	3.17572E+01	8.82443E-01	8.70662E-04	0.00000E+00	0.00000E+00
757	9.05284E-01	3.17965E+01	8.82473E-01	8.70034E-04	0.00000E+00	0.00000E+00
758	8.35064E-01	3.18358E+01	8.82410E-01	8.71142E-04	0.00000E+00	0.00000E+00
759	9.00576E-01	3.18743E+01	8.82434E-01	8.70322E-04	0.00000E+00	0.00000E+00
760	8.87853E-01	3.19155E+01	8.82442E-01	8.69202E-04	0.00000E+00	0.00000E+00
761	8.72461E-01	3.19557E+01	8.82428E-01	8.68156E-04	0.00000E+00	0.00000E+00
762	8.85743E-01	3.19952E+01	8.82433E-01	8.67024E-04	0.00000E+00	0.00000E+00
763	8.73777E-01	3.20345E+01	8.82421E-01	8.65958E-04	0.00000E+00	0.00000E+00
764	9.05848E-01	3.20720E+01	8.82452E-01	8.65368E-04	0.00000E+00	0.00000E+00
765	8.88970E-01	3.21123E+01	8.82461E-01	8.64275E-04	0.00000E+00	0.00000E+00
766	8.86938E-01	3.21517E+01	8.82467E-01	8.63163E-04	0.00000E+00	0.00000E+00
767	8.62881E-01	3.21918E+01	8.82441E-01	8.62414E-04	0.00000E+00	0.00000E+00
768	8.64585E-01	3.22303E+01	8.82418E-01	8.61603E-04	0.00000E+00	0.00000E+00
769	8.93415E-01	3.22715E+01	8.82432E-01	8.60598E-04	0.00000E+00	0.00000E+00
770	9.08474E-01	3.23118E+01	8.82466E-01	8.60145E-04	0.00000E+00	0.00000E+00
771	8.86524E-01	3.23522E+01	8.82471E-01	8.59042E-04	0.00000E+00	0.00000E+00
772	9.25830E-01	3.23915E+01	8.82528E-01	8.59772E-04	0.00000E+00	0.00000E+00
773	8.63911E-01	3.24318E+01	8.82503E-01	8.58995E-04	0.00000E+00	0.00000E+00
774	8.67589E-01	3.24712E+01	8.82484E-01	8.58100E-04	0.00000E+00	0.00000E+00
775	9.13670E-01	3.25105E+01	8.82524E-01	8.57938E-04	0.00000E+00	0.00000E+00
776	9.01353E-01	3.25498E+01	8.82549E-01	8.57174E-04	0.00000E+00	0.00000E+00
777	8.82232E-01	3.25892E+01	8.82548E-01	8.56067E-04	0.00000E+00	0.00000E+00
778	9.19355E-01	3.26303E+01	8.82596E-01	8.56278E-04	0.00000E+00	0.00000E+00
779	9.06316E-01	3.26698E+01	8.82626E-01	8.55720E-04	0.00000E+00	0.00000E+00
780	9.08899E-01	3.27100E+01	8.82660E-01	8.55286E-04	0.00000E+00	0.00000E+00
781	8.49661E-01	3.27493E+01	8.82618E-01	8.55238E-04	0.00000E+00	0.00000E+00
782	9.16196E-01	3.27888E+01	8.82661E-01	8.55224E-04	0.00000E+00	0.00000E+00
783	8.94910E-01	3.28290E+01	8.82676E-01	8.54273E-04	0.00000E+00	0.00000E+00
784	8.66590E-01	3.28693E+01	8.82656E-01	8.53428E-04	0.00000E+00	0.00000E+00
785	9.22746E-01	3.29078E+01	8.82707E-01	8.53873E-04	0.00000E+00	0.00000E+00
786	9.06284E-01	3.29490E+01	8.82737E-01	8.53314E-04	0.00000E+00	0.00000E+00
787	9.42388E-01	3.29875E+01	8.82813E-01	8.55607E-04	0.00000E+00	0.00000E+00
788	9.02136E-01	3.30268E+01	8.82838E-01	8.54871E-04	0.00000E+00	0.00000E+00
789	9.14190E-01	3.30662E+01	8.82878E-01	8.54713E-04	0.00000E+00	0.00000E+00
790	8.85970E-01	3.31055E+01	8.82881E-01	8.53637E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

791	8.49674E-01	3.31448E+01	8.82839E-01	8.53593E-04	0.00000E+00	0.00000E+00
792	9.12428E-01	3.31833E+01	8.82877E-01	8.53334E-04	0.00000E+00	0.00000E+00
793	8.72720E-01	3.32237E+01	8.82864E-01	8.52351E-04	0.00000E+00	0.00000E+00
794	8.34076E-01	3.32630E+01	8.82802E-01	8.53500E-04	0.00000E+00	0.00000E+00
795	8.64122E-01	3.33023E+01	8.82779E-01	8.52748E-04	0.00000E+00	0.00000E+00
796	8.74886E-01	3.33435E+01	8.82769E-01	8.51732E-04	0.00000E+00	0.00000E+00
797	8.48514E-01	3.33828E+01	8.82726E-01	8.51750E-04	0.00000E+00	0.00000E+00
798	8.59517E-01	3.34232E+01	8.82697E-01	8.51179E-04	0.00000E+00	0.00000E+00
799	8.55769E-01	3.34617E+01	8.82663E-01	8.50782E-04	0.00000E+00	0.00000E+00
800	8.80327E-01	3.35010E+01	8.82660E-01	8.49720E-04	0.00000E+00	0.00000E+00
801	8.92688E-01	3.35412E+01	8.82672E-01	8.48748E-04	0.00000E+00	0.00000E+00
802	8.80447E-01	3.35815E+01	8.82670E-01	8.47691E-04	0.00000E+00	0.00000E+00
803	8.65536E-01	3.36208E+01	8.82648E-01	8.46903E-04	0.00000E+00	0.00000E+00
804	8.77887E-01	3.36612E+01	8.82642E-01	8.45867E-04	0.00000E+00	0.00000E+00
805	8.66415E-01	3.37005E+01	8.82622E-01	8.45055E-04	0.00000E+00	0.00000E+00
806	8.55633E-01	3.37398E+01	8.82589E-01	8.44670E-04	0.00000E+00	0.00000E+00
807	8.76156E-01	3.37802E+01	8.82581E-01	8.43658E-04	0.00000E+00	0.00000E+00
808	8.65543E-01	3.38187E+01	8.82559E-01	8.42876E-04	0.00000E+00	0.00000E+00
809	8.98443E-01	3.38598E+01	8.82579E-01	8.42061E-04	0.00000E+00	0.00000E+00
810	8.72591E-01	3.38992E+01	8.82567E-01	8.41109E-04	0.00000E+00	0.00000E+00
811	8.63937E-01	3.39395E+01	8.82544E-01	8.40384E-04	0.00000E+00	0.00000E+00
812	8.59286E-01	3.39807E+01	8.82515E-01	8.39837E-04	0.00000E+00	0.00000E+00
813	8.92750E-01	3.40200E+01	8.82528E-01	8.38896E-04	0.00000E+00	0.00000E+00
814	8.70065E-01	3.40603E+01	8.82512E-01	8.38002E-04	0.00000E+00	0.00000E+00
815	9.06882E-01	3.41005E+01	8.82542E-01	8.37508E-04	0.00000E+00	0.00000E+00
816	9.02221E-01	3.41390E+01	8.82566E-01	8.36827E-04	0.00000E+00	0.00000E+00
817	8.79265E-01	3.41783E+01	8.82562E-01	8.35810E-04	0.00000E+00	0.00000E+00
818	9.30025E-01	3.42177E+01	8.82621E-01	8.36809E-04	0.00000E+00	0.00000E+00
819	8.94065E-01	3.42572E+01	8.82635E-01	8.35901E-04	0.00000E+00	0.00000E+00
820	9.15666E-01	3.42965E+01	8.82675E-01	8.35855E-04	0.00000E+00	0.00000E+00
821	8.91504E-01	3.43368E+01	8.82686E-01	8.34903E-04	0.00000E+00	0.00000E+00
822	8.93009E-01	3.43770E+01	8.82698E-01	8.33979E-04	0.00000E+00	0.00000E+00
823	8.86628E-01	3.44163E+01	8.82703E-01	8.32977E-04	0.00000E+00	0.00000E+00
824	8.61377E-01	3.44777E+01	8.82677E-01	8.32367E-04	0.00000E+00	0.00000E+00
825	8.75953E-01	3.45217E+01	8.82669E-01	8.31395E-04	0.00000E+00	0.00000E+00
826	8.70027E-01	3.45620E+01	8.82654E-01	8.30527E-04	0.00000E+00	0.00000E+00
827	8.77346E-01	3.46003E+01	8.82647E-01	8.29545E-04	0.00000E+00	0.00000E+00
828	8.80828E-01	3.46398E+01	8.82645E-01	8.28543E-04	0.00000E+00	0.00000E+00
829	8.50566E-01	3.46800E+01	8.82606E-01	8.28449E-04	0.00000E+00	0.00000E+00
830	8.81076E-01	3.47193E+01	8.82604E-01	8.27450E-04	0.00000E+00	0.00000E+00
831	8.42050E-01	3.47588E+01	8.82555E-01	8.27898E-04	0.00000E+00	0.00000E+00
832	8.87007E-01	3.47982E+01	8.82561E-01	8.26917E-04	0.00000E+00	0.00000E+00
833	9.30243E-01	3.48365E+01	8.82618E-01	8.27912E-04	0.00000E+00	0.00000E+00
834	8.80511E-01	3.48760E+01	8.82616E-01	8.26920E-04	0.00000E+00	0.00000E+00
835	8.96821E-01	3.49162E+01	8.82633E-01	8.26103E-04	0.00000E+00	0.00000E+00
836	8.68988E-01	3.49555E+01	8.82616E-01	8.25274E-04	0.00000E+00	0.00000E+00
837	8.71200E-01	3.49950E+01	8.82603E-01	8.24399E-04	0.00000E+00	0.00000E+00
838	8.90257E-01	3.50343E+01	8.82612E-01	8.23463E-04	0.00000E+00	0.00000E+00
839	8.66278E-01	3.50737E+01	8.82592E-01	8.22710E-04	0.00000E+00	0.00000E+00
840	8.88377E-01	3.51122E+01	8.82599E-01	8.21757E-04	0.00000E+00	0.00000E+00
841	8.82194E-01	3.51515E+01	8.82599E-01	8.20777E-04	0.00000E+00	0.00000E+00
842	8.79707E-01	3.51908E+01	8.82595E-01	8.19806E-04	0.00000E+00	0.00000E+00
843	9.10384E-01	3.52293E+01	8.82628E-01	8.19497E-04	0.00000E+00	0.00000E+00
844	8.94507E-01	3.52687E+01	8.82642E-01	8.18645E-04	0.00000E+00	0.00000E+00
845	8.90709E-01	3.53098E+01	8.82652E-01	8.17729E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

846	8.69568E-01	3.53502E+01	8.82637E-01	8.16907E-04	0.00000E+00	0.00000E+00
847	8.87264E-01	3.53895E+01	8.82642E-01	8.15958E-04	0.00000E+00	0.00000E+00
848	8.94858E-01	3.54280E+01	8.82656E-01	8.15121E-04	0.00000E+00	0.00000E+00
849	8.37074E-01	3.54663E+01	8.82603E-01	8.15935E-04	0.00000E+00	0.00000E+00
850	9.13930E-01	3.55067E+01	8.82640E-01	8.15809E-04	0.00000E+00	0.00000E+00
851	8.79032E-01	3.55460E+01	8.82635E-01	8.14858E-04	0.00000E+00	0.00000E+00
852	8.85122E-01	3.55845E+01	8.82638E-01	8.13904E-04	0.00000E+00	0.00000E+00
853	8.94571E-01	3.56248E+01	8.82652E-01	8.13068E-04	0.00000E+00	0.00000E+00
854	9.62724E-01	3.56642E+01	8.82746E-01	8.17533E-04	0.00000E+00	0.00000E+00
855	8.82731E-01	3.57043E+01	8.82746E-01	8.16574E-04	0.00000E+00	0.00000E+00
856	8.89285E-01	3.57428E+01	8.82754E-01	8.15654E-04	0.00000E+00	0.00000E+00
857	9.23743E-01	3.57822E+01	8.82802E-01	8.16108E-04	0.00000E+00	0.00000E+00
858	8.76216E-01	3.58225E+01	8.82794E-01	8.15191E-04	0.00000E+00	0.00000E+00
859	8.61300E-01	3.58618E+01	8.82769E-01	8.14625E-04	0.00000E+00	0.00000E+00
860	8.91294E-01	3.59003E+01	8.82779E-01	8.13736E-04	0.00000E+00	0.00000E+00
861	9.30004E-01	3.59397E+01	8.82834E-01	8.14645E-04	0.00000E+00	0.00000E+00
862	9.32903E-01	3.59790E+01	8.82892E-01	8.15777E-04	0.00000E+00	0.00000E+00
863	8.97895E-01	3.60193E+01	8.82910E-01	8.15016E-04	0.00000E+00	0.00000E+00
864	9.10402E-01	3.60587E+01	8.82942E-01	8.14694E-04	0.00000E+00	0.00000E+00
865	8.65754E-01	3.60990E+01	8.82922E-01	8.13993E-04	0.00000E+00	0.00000E+00
866	8.69823E-01	3.61392E+01	8.82906E-01	8.13192E-04	0.00000E+00	0.00000E+00
867	8.88938E-01	3.61795E+01	8.82913E-01	8.12281E-04	0.00000E+00	0.00000E+00
868	9.00093E-01	3.62198E+01	8.82933E-01	8.11585E-04	0.00000E+00	0.00000E+00
869	9.15167E-01	3.62592E+01	8.82970E-01	8.11501E-04	0.00000E+00	0.00000E+00
870	8.96414E-01	3.62995E+01	8.82986E-01	8.10713E-04	0.00000E+00	0.00000E+00
871	8.49652E-01	3.63378E+01	8.82948E-01	8.10688E-04	0.00000E+00	0.00000E+00
872	9.15365E-01	3.63790E+01	8.82985E-01	8.10612E-04	0.00000E+00	0.00000E+00
873	8.71667E-01	3.64193E+01	8.82972E-01	8.09785E-04	0.00000E+00	0.00000E+00
874	9.00009E-01	3.64587E+01	8.82991E-01	8.09092E-04	0.00000E+00	0.00000E+00
875	8.50244E-01	3.64980E+01	8.82954E-01	8.09035E-04	0.00000E+00	0.00000E+00
876	8.84192E-01	3.65383E+01	8.82955E-01	8.08110E-04	0.00000E+00	0.00000E+00
877	8.94833E-01	3.65787E+01	8.82969E-01	8.07300E-04	0.00000E+00	0.00000E+00
878	8.59358E-01	3.66180E+01	8.82942E-01	8.06828E-04	0.00000E+00	0.00000E+00
879	8.80722E-01	3.66592E+01	8.82939E-01	8.05912E-04	0.00000E+00	0.00000E+00
880	8.72658E-01	3.66985E+01	8.82928E-01	8.05078E-04	0.00000E+00	0.00000E+00
881	8.36667E-01	3.67378E+01	8.82875E-01	8.05882E-04	0.00000E+00	0.00000E+00
882	8.81112E-01	3.67773E+01	8.82873E-01	8.04968E-04	0.00000E+00	0.00000E+00
883	8.93337E-01	3.68167E+01	8.82885E-01	8.04142E-04	0.00000E+00	0.00000E+00
884	8.95296E-01	3.68560E+01	8.82899E-01	8.03353E-04	0.00000E+00	0.00000E+00
885	8.84254E-01	3.68953E+01	8.82901E-01	8.02444E-04	0.00000E+00	0.00000E+00
886	9.00281E-01	3.69347E+01	8.82920E-01	8.01777E-04	0.00000E+00	0.00000E+00
887	8.85138E-01	3.69760E+01	8.82923E-01	8.00874E-04	0.00000E+00	0.00000E+00
888	9.13759E-01	3.70172E+01	8.82957E-01	8.00727E-04	0.00000E+00	0.00000E+00
889	8.64624E-01	3.70573E+01	8.82937E-01	8.00090E-04	0.00000E+00	0.00000E+00
890	8.92166E-01	3.70968E+01	8.82947E-01	7.99257E-04	0.00000E+00	0.00000E+00
891	8.66125E-01	3.71362E+01	8.82928E-01	7.98581E-04	0.00000E+00	0.00000E+00
892	9.02822E-01	3.71755E+01	8.82951E-01	7.97997E-04	0.00000E+00	0.00000E+00
893	8.68450E-01	3.72158E+01	8.82934E-01	7.97266E-04	0.00000E+00	0.00000E+00
894	8.92813E-01	3.72552E+01	8.82945E-01	7.96449E-04	0.00000E+00	0.00000E+00
895	8.64901E-01	3.72927E+01	8.82925E-01	7.95813E-04	0.00000E+00	0.00000E+00
896	8.41329E-01	3.73330E+01	8.82879E-01	7.96283E-04	0.00000E+00	0.00000E+00
897	8.45436E-01	3.73732E+01	8.82837E-01	7.96493E-04	0.00000E+00	0.00000E+00
898	9.02403E-01	3.74135E+01	8.82859E-01	7.95903E-04	0.00000E+00	0.00000E+00
899	8.89752E-01	3.74520E+01	8.82866E-01	7.95052E-04	0.00000E+00	0.00000E+00
900	8.89794E-01	3.74913E+01	8.82874E-01	7.94204E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

901	8.60332E-01	3.75298E+01	8.82849E-01	7.93716E-04	0.00000E+00	0.00000E+00
902	9.30763E-01	3.75710E+01	8.82902E-01	7.94619E-04	0.00000E+00	0.00000E+00
903	8.80199E-01	3.76112E+01	8.82899E-01	7.93742E-04	0.00000E+00	0.00000E+00
904	9.03355E-01	3.76515E+01	8.82922E-01	7.93186E-04	0.00000E+00	0.00000E+00
905	8.98741E-01	3.76945E+01	8.82939E-01	7.92501E-04	0.00000E+00	0.00000E+00
906	8.96136E-01	3.77338E+01	8.82954E-01	7.91758E-04	0.00000E+00	0.00000E+00
907	8.69383E-01	3.77733E+01	8.82939E-01	7.91025E-04	0.00000E+00	0.00000E+00
908	8.44234E-01	3.78127E+01	8.82896E-01	7.91305E-04	0.00000E+00	0.00000E+00
909	8.63161E-01	3.78510E+01	8.82875E-01	7.90732E-04	0.00000E+00	0.00000E+00
910	8.81043E-01	3.78905E+01	8.82873E-01	7.89863E-04	0.00000E+00	0.00000E+00
911	8.85773E-01	3.79317E+01	8.82876E-01	7.89000E-04	0.00000E+00	0.00000E+00
912	8.78173E-01	3.79700E+01	8.82871E-01	7.88150E-04	0.00000E+00	0.00000E+00
913	8.76063E-01	3.80103E+01	8.82863E-01	7.87320E-04	0.00000E+00	0.00000E+00
914	9.00040E-01	3.80497E+01	8.82882E-01	7.86681E-04	0.00000E+00	0.00000E+00
915	8.73214E-01	3.80900E+01	8.82871E-01	7.85891E-04	0.00000E+00	0.00000E+00
916	8.86104E-01	3.81293E+01	8.82875E-01	7.85038E-04	0.00000E+00	0.00000E+00
917	8.75636E-01	3.81687E+01	8.82867E-01	7.84220E-04	0.00000E+00	0.00000E+00
918	8.84182E-01	3.82072E+01	8.82868E-01	7.83364E-04	0.00000E+00	0.00000E+00
919	9.22498E-01	3.82475E+01	8.82912E-01	7.83702E-04	0.00000E+00	0.00000E+00
920	8.96132E-01	3.82877E+01	8.82926E-01	7.82980E-04	0.00000E+00	0.00000E+00
921	9.25559E-01	3.83280E+01	8.82972E-01	7.83503E-04	0.00000E+00	0.00000E+00
922	8.97636E-01	3.83673E+01	8.82988E-01	7.82813E-04	0.00000E+00	0.00000E+00
923	8.52250E-01	3.84067E+01	8.82955E-01	7.82674E-04	0.00000E+00	0.00000E+00
924	9.24079E-01	3.84452E+01	8.83000E-01	7.83096E-04	0.00000E+00	0.00000E+00
925	8.83102E-01	3.84855E+01	8.83000E-01	7.82247E-04	0.00000E+00	0.00000E+00
926	8.92250E-01	3.85257E+01	8.83010E-01	7.81464E-04	0.00000E+00	0.00000E+00
927	9.02516E-01	3.85642E+01	8.83031E-01	7.80904E-04	0.00000E+00	0.00000E+00
928	8.62153E-01	3.86035E+01	8.83008E-01	7.80386E-04	0.00000E+00	0.00000E+00
929	9.16951E-01	3.86447E+01	8.83045E-01	7.80403E-04	0.00000E+00	0.00000E+00
930	8.92503E-01	3.86842E+01	8.83055E-01	7.79628E-04	0.00000E+00	0.00000E+00
931	8.78143E-01	3.87235E+01	8.83050E-01	7.78806E-04	0.00000E+00	0.00000E+00
932	8.79195E-01	3.87628E+01	8.83046E-01	7.77980E-04	0.00000E+00	0.00000E+00
933	8.97217E-01	3.88032E+01	8.83061E-01	7.77293E-04	0.00000E+00	0.00000E+00
934	8.88467E-01	3.88425E+01	8.83067E-01	7.76480E-04	0.00000E+00	0.00000E+00
935	8.92736E-01	3.88818E+01	8.83077E-01	7.75716E-04	0.00000E+00	0.00000E+00
936	8.37429E-01	3.89212E+01	8.83028E-01	7.76425E-04	0.00000E+00	0.00000E+00
937	8.99421E-01	3.89623E+01	8.83046E-01	7.75792E-04	0.00000E+00	0.00000E+00
938	9.14554E-01	3.90018E+01	8.83079E-01	7.75694E-04	0.00000E+00	0.00000E+00
939	8.58625E-01	3.90412E+01	8.83053E-01	7.75305E-04	0.00000E+00	0.00000E+00
940	8.47818E-01	3.90805E+01	8.83016E-01	7.75389E-04	0.00000E+00	0.00000E+00
941	8.56232E-01	3.91208E+01	8.82987E-01	7.75087E-04	0.00000E+00	0.00000E+00
942	9.31864E-01	3.91610E+01	8.83039E-01	7.76006E-04	0.00000E+00	0.00000E+00
943	9.12862E-01	3.92003E+01	8.83071E-01	7.75829E-04	0.00000E+00	0.00000E+00
944	8.79352E-01	3.92398E+01	8.83067E-01	7.75015E-04	0.00000E+00	0.00000E+00
945	8.88809E-01	3.92782E+01	8.83073E-01	7.74216E-04	0.00000E+00	0.00000E+00
946	8.97479E-01	3.93175E+01	8.83088E-01	7.73546E-04	0.00000E+00	0.00000E+00
947	8.66574E-01	3.93578E+01	8.83071E-01	7.72925E-04	0.00000E+00	0.00000E+00
948	9.03253E-01	3.93982E+01	8.83092E-01	7.72402E-04	0.00000E+00	0.00000E+00
949	9.01569E-01	3.94375E+01	8.83112E-01	7.71833E-04	0.00000E+00	0.00000E+00
950	8.75947E-01	3.94778E+01	8.83104E-01	7.71055E-04	0.00000E+00	0.00000E+00
951	8.75460E-01	3.95190E+01	8.83096E-01	7.70285E-04	0.00000E+00	0.00000E+00
952	8.95944E-01	3.95575E+01	8.83110E-01	7.69592E-04	0.00000E+00	0.00000E+00
953	8.96185E-01	3.95968E+01	8.83123E-01	7.68905E-04	0.00000E+00	0.00000E+00
954	8.74146E-01	3.96352E+01	8.83114E-01	7.68155E-04	0.00000E+00	0.00000E+00
955	9.05963E-01	3.96755E+01	8.83138E-01	7.67723E-04	0.00000E+00	0.00000E+00

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

956	9.06859E-01	3.97140E+01	8.83163E-01	7.67321E-04	0.00000E+00	0.00000E+00
957	8.83999E-01	3.97542E+01	8.83164E-01	7.66518E-04	0.00000E+00	0.00000E+00
958	9.09150E-01	3.97937E+01	8.83191E-01	7.66198E-04	0.00000E+00	0.00000E+00
959	8.57473E-01	3.98338E-01	8.83164E-01	7.65868E-04	0.00000E+00	0.00000E+00
960	8.94787E-01	3.98732E+01	8.83176E-01	7.65165E-04	0.00000E+00	0.00000E+00
961	8.97012E-01	3.99135E+01	8.83190E-01	7.64502E-04	0.00000E+00	0.00000E+00
962	9.30624E-01	3.99538E+01	8.83240E-01	7.65302E-04	0.00000E+00	0.00000E+00
963	8.89713E-01	3.99940E+01	8.83247E-01	7.64535E-04	0.00000E+00	0.00000E+00
964	8.49902E-01	4.00335E+01	8.83212E-01	7.64526E-04	0.00000E+00	0.00000E+00
965	8.76986E-01	4.00728E+01	8.83205E-01	7.63759E-04	0.00000E+00	0.00000E+00
966	8.78277E-01	4.01122E+01	8.83200E-01	7.62984E-04	0.00000E+00	0.00000E+00
967	8.85869E-01	4.01507E+01	8.83203E-01	7.62198E-04	0.00000E+00	0.00000E+00
968	8.87499E-01	4.01900E+01	8.83208E-01	7.61421E-04	0.00000E+00	0.00000E+00
969	8.49915E-01	4.02293E+01	8.83173E-01	7.61412E-04	0.00000E+00	0.00000E+00
970	8.78878E-01	4.02687E+01	8.83169E-01	7.60638E-04	0.00000E+00	0.00000E+00
971	8.67161E-01	4.03072E+01	8.83152E-01	7.60032E-04	0.00000E+00	0.00000E+00
972	8.86124E-01	4.03483E+01	8.83155E-01	7.59255E-04	0.00000E+00	0.00000E+00
973	8.91461E-01	4.03877E+01	8.83164E-01	7.58521E-04	0.00000E+00	0.00000E+00
974	9.03433E-01	4.04262E+01	8.83185E-01	7.58027E-04	0.00000E+00	0.00000E+00
975	8.74918E-01	4.04665E+01	8.83176E-01	7.57295E-04	0.00000E+00	0.00000E+00
976	8.61610E-01	4.05067E+01	8.83154E-01	7.56841E-04	0.00000E+00	0.00000E+00
977	9.49750E-01	4.05470E+01	8.83222E-01	7.59143E-04	0.00000E+00	0.00000E+00
978	8.64457E-01	4.05873E+01	8.83203E-01	7.58609E-04	0.00000E+00	0.00000E+00
979	8.95833E-01	4.06275E+01	8.83216E-01	7.57942E-04	0.00000E+00	0.00000E+00
980	9.00901E-01	4.06678E+01	8.83234E-01	7.57383E-04	0.00000E+00	0.00000E+00
981	9.08891E-01	4.07063E+01	8.83260E-01	7.57062E-04	0.00000E+00	0.00000E+00
982	8.83295E-01	4.07457E+01	8.83260E-01	7.56290E-04	0.00000E+00	0.00000E+00
983	8.70552E-01	4.07860E+01	8.83247E-01	7.55629E-04	0.00000E+00	0.00000E+00
984	9.07649E-01	4.08262E+01	8.83272E-01	7.55268E-04	0.00000E+00	0.00000E+00
985	9.11443E-01	4.08655E+01	8.83301E-01	7.55044E-04	0.00000E+00	0.00000E+00
986	8.71276E-01	4.09050E+01	8.83289E-01	7.54375E-04	0.00000E+00	0.00000E+00
987	8.49950E-01	4.09443E+01	8.83255E-01	7.54368E-04	0.00000E+00	0.00000E+00
988	8.96020E-01	4.09837E+01	8.83268E-01	7.53714E-04	0.00000E+00	0.00000E+00
989	8.92620E-01	4.10248E+01	8.83277E-01	7.53010E-04	0.00000E+00	0.00000E+00
990	9.06883E-01	4.10660E+01	8.83301E-01	7.52626E-04	0.00000E+00	0.00000E+00
991	8.86281E-01	4.11063E+01	8.83304E-01	7.51871E-04	0.00000E+00	0.00000E+00
992	8.75118E-01	4.11467E+01	8.83296E-01	7.51157E-04	0.00000E+00	0.00000E+00
993	8.92430E-01	4.11860E+01	8.83305E-01	7.50455E-04	0.00000E+00	0.00000E+00
994	9.04477E-01	4.12262E+01	8.83326E-01	7.50002E-04	0.00000E+00	0.00000E+00
995	8.64342E-01	4.12665E+01	8.83307E-01	7.49490E-04	0.00000E+00	0.00000E+00
996	8.91975E-01	4.13068E+01	8.83316E-01	7.48786E-04	0.00000E+00	0.00000E+00
997	9.12615E-01	4.13462E+01	8.83345E-01	7.48613E-04	0.00000E+00	0.00000E+00
998	8.50638E-01	4.13855E+01	8.83313E-01	7.48581E-04	0.00000E+00	0.00000E+00
999	9.06102E-01	4.14258E+01	8.83336E-01	7.48180E-04	0.00000E+00	0.00000E+00
1000	8.96657E-01	4.14652E+01	8.83349E-01	7.47549E-04	0.00000E+00	0.00000E+00
1001	8.77960E-01	4.15055E+01	8.83343E-01	7.46819E-04	0.00000E+00	0.00000E+00
1002	9.06259E-01	4.15457E+01	8.83366E-01	7.46424E-04	0.00000E+00	0.00000E+00
1003	8.59517E-01	4.15852E+01	8.83343E-01	7.46059E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123

EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)							
LIFETIME = 3.48455E-05 + OR - 6.93436E-08		GENERATION TIME = 2.47240E-05 + OR - 3.63759E-08					
NU BAR = 2.44275E+00 + OR - 6.92320E-05		AVERAGE FISSION GROUP = 2.16236E+01 + OR - 3.91568E-03					
		ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 3.01215E-01 + OR - 9.58153E-04					
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES	
3	0.88338	+ OR - 0.00075	0.88264 TO 0.88413	0.88189 TO 0.88487	0.88114 TO 0.88562	1000000	
4	0.88336	+ OR - 0.00075	0.88261 TO 0.88411	0.88187 TO 0.88485	0.88112 TO 0.88560	999000	
5	0.88332	+ OR - 0.00075	0.88257 TO 0.88406	0.88183 TO 0.88481	0.88108 TO 0.88556	998000	
6	0.88335	+ OR - 0.00075	0.88260 TO 0.88410	0.88186 TO 0.88484	0.88111 TO 0.88559	997000	
7	0.88331	+ OR - 0.00075	0.88257 TO 0.88406	0.88182 TO 0.88480	0.88107 TO 0.88555	996000	
8	0.88332	+ OR - 0.00075	0.88257 TO 0.88407	0.88183 TO 0.88481	0.88108 TO 0.88556	995000	
9	0.88329	+ OR - 0.00075	0.88255 TO 0.88404	0.88180 TO 0.88479	0.88105 TO 0.88553	994000	
10	0.88333	+ OR - 0.00075	0.88258 TO 0.88408	0.88184 TO 0.88482	0.88109 TO 0.88557	993000	
11	0.88331	+ OR - 0.00075	0.88256 TO 0.88406	0.88182 TO 0.88480	0.88107 TO 0.88555	992000	
12	0.88330	+ OR - 0.00075	0.88255 TO 0.88405	0.88181 TO 0.88480	0.88106 TO 0.88555	991000	
17	0.88338	+ OR - 0.00075	0.88264 TO 0.88413	0.88189 TO 0.88488	0.88114 TO 0.88563	986000	
22	0.88338	+ OR - 0.00075	0.88263 TO 0.88413	0.88188 TO 0.88489	0.88112 TO 0.88564	981000	
27	0.88350	+ OR - 0.00075	0.88275 TO 0.88426	0.88200 TO 0.88501	0.88124 TO 0.88576	976000	
32	0.88358	+ OR - 0.00076	0.88283 TO 0.88434	0.88207 TO 0.88510	0.88132 TO 0.88585	971000	
37	0.88367	+ OR - 0.00076	0.88291 TO 0.88442	0.88215 TO 0.88518	0.88139 TO 0.88594	966000	
42	0.88366	+ OR - 0.00076	0.88290 TO 0.88442	0.88214 TO 0.88518	0.88138 TO 0.88594	961000	
47	0.88377	+ OR - 0.00076	0.88301 TO 0.88453	0.88225 TO 0.88529	0.88149 TO 0.88605	956000	
52	0.88387	+ OR - 0.00076	0.88311 TO 0.88463	0.88234 TO 0.88539	0.88158 TO 0.88616	951000	
57	0.88387	+ OR - 0.00077	0.88311 TO 0.88464	0.88234 TO 0.88540	0.88158 TO 0.88617	946000	
62	0.88386	+ OR - 0.00077	0.88309 TO 0.88462	0.88232 TO 0.88539	0.88155 TO 0.88616	941000	

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

67	0.88390	+ OR - 0.00077	0.88313 TO 0.88467	0.88236 TO 0.88544	0.88159 TO 0.88621	936000
72	0.88391	+ OR - 0.00077	0.88314 TO 0.88468	0.88237 TO 0.88546	0.88160 TO 0.88623	931000
77	0.88390	+ OR - 0.00077	0.88313 TO 0.88468	0.88235 TO 0.88545	0.88158 TO 0.88623	926000
82	0.88412	+ OR - 0.00077	0.88335 TO 0.88489	0.88258 TO 0.88566	0.88181 TO 0.88643	921000
87	0.88413	+ OR - 0.00077	0.88335 TO 0.88490	0.88258 TO 0.88567	0.88181 TO 0.88645	916000
92	0.88415	+ OR - 0.00077	0.88338 TO 0.88493	0.88260 TO 0.88570	0.88183 TO 0.88648	911000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
97	0.88418	+ OR - 0.00077	0.88341 TO 0.88496	0.88263 TO 0.88573	0.88186 TO 0.88650	906000
102	0.88421	+ OR - 0.00077	0.88344 TO 0.88498	0.88266 TO 0.88575	0.88189 TO 0.88653	901000
107	0.88418	+ OR - 0.00077	0.88341 TO 0.88496	0.88263 TO 0.88573	0.88186 TO 0.88650	896000
112	0.88407	+ OR - 0.00077	0.88330 TO 0.88485	0.88253 TO 0.88562	0.88176 TO 0.88639	891000
117	0.88420	+ OR - 0.00077	0.88343 TO 0.88497	0.88266 TO 0.88574	0.88189 TO 0.88652	886000
122	0.88431	+ OR - 0.00077	0.88354 TO 0.88508	0.88276 TO 0.88585	0.88199 TO 0.88662	881000
127	0.88431	+ OR - 0.00078	0.88354 TO 0.88509	0.88276 TO 0.88587	0.88199 TO 0.88664	876000
132	0.88432	+ OR - 0.00078	0.88354 TO 0.88510	0.88276 TO 0.88588	0.88198 TO 0.88666	871000
137	0.88417	+ OR - 0.00078	0.88339 TO 0.88495	0.88261 TO 0.88573	0.88183 TO 0.88651	866000
142	0.88428	+ OR - 0.00078	0.88350 TO 0.88506	0.88272 TO 0.88584	0.88194 TO 0.88662	861000
147	0.88428	+ OR - 0.00078	0.88349 TO 0.88506	0.88271 TO 0.88585	0.88193 TO 0.88663	856000
152	0.88424	+ OR - 0.00079	0.88345 TO 0.88503	0.88266 TO 0.88581	0.88188 TO 0.88660	851000
157	0.88418	+ OR - 0.00079	0.88339 TO 0.88497	0.88260 TO 0.88576	0.88181 TO 0.88655	846000
162	0.88417	+ OR - 0.00079	0.88337 TO 0.88496	0.88258 TO 0.88575	0.88179 TO 0.88654	841000
167	0.88418	+ OR - 0.00080	0.88338 TO 0.88497	0.88259 TO 0.88577	0.88179 TO 0.88656	836000
172	0.88427	+ OR - 0.00080	0.88347 TO 0.88507	0.88267 TO 0.88587	0.88188 TO 0.88666	831000
177	0.88428	+ OR - 0.00080	0.88348 TO 0.88508	0.88268 TO 0.88588	0.88188 TO 0.88668	826000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

182	0.88420	+ OR - 0.00080	0.88339 TO 0.88500	0.88259 TO 0.88580	0.88179 TO 0.88660	821000
187	0.88421	+ OR - 0.00081	0.88341 TO 0.88502	0.88260 TO 0.88582	0.88180 TO 0.88663	816000
192	0.88430	+ OR - 0.00081	0.88349 TO 0.88511	0.88268 TO 0.88592	0.88187 TO 0.88672	811000
197	0.88432	+ OR - 0.00081	0.88350 TO 0.88513	0.88269 TO 0.88594	0.88188 TO 0.88675	806000
202	0.88422	+ OR - 0.00082	0.88340 TO 0.88504	0.88259 TO 0.88585	0.88177 TO 0.88667	801000
207	0.88415	+ OR - 0.00082	0.88333 TO 0.88496	0.88251 TO 0.88578	0.88169 TO 0.88660	796000
212	0.88414	+ OR - 0.00082	0.88332 TO 0.88496	0.88250 TO 0.88579	0.88167 TO 0.88661	791000
217	0.88415	+ OR - 0.00082	0.88332 TO 0.88497	0.88250 TO 0.88579	0.88168 TO 0.88662	786000
222	0.88404	+ OR - 0.00082	0.88322 TO 0.88487	0.88239 TO 0.88569	0.88157 TO 0.88652	781000
227	0.88408	+ OR - 0.00083	0.88325 TO 0.88491	0.88242 TO 0.88574	0.88159 TO 0.88657	776000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
232	0.88419	+ OR - 0.00083	0.88336 TO 0.88502	0.88252 TO 0.88585	0.88169 TO 0.88669	771000
237	0.88425	+ OR - 0.00083	0.88342 TO 0.88509	0.88258 TO 0.88592	0.88175 TO 0.88676	766000
242	0.88418	+ OR - 0.00083	0.88335 TO 0.88501	0.88252 TO 0.88584	0.88168 TO 0.88668	761000
247	0.88425	+ OR - 0.00084	0.88341 TO 0.88508	0.88258 TO 0.88592	0.88174 TO 0.88675	756000
252	0.88421	+ OR - 0.00084	0.88337 TO 0.88504	0.88253 TO 0.88588	0.88170 TO 0.88672	751000
257	0.88431	+ OR - 0.00084	0.88347 TO 0.88515	0.88263 TO 0.88600	0.88179 TO 0.88684	746000
262	0.88422	+ OR - 0.00084	0.88338 TO 0.88507	0.88254 TO 0.88591	0.88170 TO 0.88675	741000
267	0.88418	+ OR - 0.00085	0.88333 TO 0.88503	0.88248 TO 0.88587	0.88164 TO 0.88672	736000
272	0.88431	+ OR - 0.00085	0.88346 TO 0.88515	0.88261 TO 0.88600	0.88176 TO 0.88685	731000
277	0.88427	+ OR - 0.00085	0.88342 TO 0.88513	0.88257 TO 0.88598	0.88172 TO 0.88683	726000
282	0.88425	+ OR - 0.00086	0.88339 TO 0.88510	0.88254 TO 0.88596	0.88168 TO 0.88681	721000
287	0.88427	+ OR - 0.00086	0.88341 TO 0.88513	0.88255 TO 0.88599	0.88169 TO 0.88685	716000
292	0.88429	+ OR - 0.00086	0.88343 TO 0.88516	0.88256 TO 0.88602	0.88170 TO 0.88689	711000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

297	0.88429	+ OR - 0.00087	0.88342 TO 0.88516	0.88255 TO 0.88603	0.88168 TO 0.88690	706000
302	0.88429	+ OR - 0.00086	0.88342 TO 0.88515	0.88256 TO 0.88601	0.88169 TO 0.88688	701000
307	0.88431	+ OR - 0.00087	0.88344 TO 0.88518	0.88258 TO 0.88605	0.88171 TO 0.88692	696000
312	0.88432	+ OR - 0.00087	0.88345 TO 0.88519	0.88258 TO 0.88605	0.88172 TO 0.88692	691000
317	0.88421	+ OR - 0.00087	0.88334 TO 0.88508	0.88247 TO 0.88595	0.88160 TO 0.88681	686000
322	0.88410	+ OR - 0.00087	0.88323 TO 0.88496	0.88236 TO 0.88583	0.88149 TO 0.88670	681000
327	0.88424	+ OR - 0.00087	0.88338 TO 0.88511	0.88251 TO 0.88598	0.88164 TO 0.88685	676000
332	0.88431	+ OR - 0.00087	0.88344 TO 0.88518	0.88256 TO 0.88606	0.88169 TO 0.88693	671000
337	0.88427	+ OR - 0.00087	0.88340 TO 0.88514	0.88253 TO 0.88601	0.88165 TO 0.88689	666000
342	0.88425	+ OR - 0.00088	0.88337 TO 0.88512	0.88250 TO 0.88600	0.88162 TO 0.88687	661000
347	0.88419	+ OR - 0.00088	0.88331 TO 0.88507	0.88242 TO 0.88595	0.88154 TO 0.88683	656000
352	0.88421	+ OR - 0.00089	0.88332 TO 0.88509	0.88243 TO 0.88598	0.88154 TO 0.88687	651000
357	0.88420	+ OR - 0.00089	0.88331 TO 0.88509	0.88241 TO 0.88598	0.88152 TO 0.88688	646000
362	0.88424	+ OR - 0.00090	0.88334 TO 0.88514	0.88245 TO 0.88604	0.88155 TO 0.88693	641000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
367	0.88423	+ OR - 0.00090	0.88333 TO 0.88513	0.88242 TO 0.88603	0.88152 TO 0.88694	636000
372	0.88420	+ OR - 0.00091	0.88330 TO 0.88511	0.88239 TO 0.88602	0.88148 TO 0.88693	631000
377	0.88406	+ OR - 0.00091	0.88315 TO 0.88497	0.88224 TO 0.88588	0.88133 TO 0.88679	626000
382	0.88413	+ OR - 0.00091	0.88322 TO 0.88505	0.88230 TO 0.88596	0.88139 TO 0.88687	621000
387	0.88412	+ OR - 0.00092	0.88320 TO 0.88504	0.88228 TO 0.88596	0.88136 TO 0.88688	616000
392	0.88415	+ OR - 0.00093	0.88322 TO 0.88507	0.88230 TO 0.88600	0.88137 TO 0.88693	611000
397	0.88410	+ OR - 0.00093	0.88317 TO 0.88503	0.88224 TO 0.88596	0.88131 TO 0.88689	606000
402	0.88405	+ OR - 0.00094	0.88311 TO 0.88498	0.88218 TO 0.88592	0.88124 TO 0.88686	601000
407	0.88417	+ OR - 0.00094	0.88323 TO 0.88511	0.88229 TO 0.88605	0.88135 TO 0.88699	596000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

412	0.88413	+ OR - 0.00095	0.88318 TO 0.88508	0.88224 TO 0.88602	0.88129 TO 0.88697	591000
417	0.88408	+ OR - 0.00095	0.88313 TO 0.88503	0.88218 TO 0.88598	0.88123 TO 0.88693	586000
422	0.88411	+ OR - 0.00096	0.88315 TO 0.88506	0.88220 TO 0.88602	0.88124 TO 0.88697	581000
427	0.88403	+ OR - 0.00096	0.88307 TO 0.88500	0.88211 TO 0.88596	0.88115 TO 0.88692	576000
432	0.88407	+ OR - 0.00097	0.88311 TO 0.88504	0.88214 TO 0.88601	0.88117 TO 0.88698	571000
437	0.88402	+ OR - 0.00097	0.88305 TO 0.88499	0.88209 TO 0.88595	0.88112 TO 0.88692	566000
442	0.88409	+ OR - 0.00097	0.88312 TO 0.88505	0.88215 TO 0.88602	0.88119 TO 0.88698	561000
447	0.88417	+ OR - 0.00097	0.88320 TO 0.88514	0.88223 TO 0.88611	0.88126 TO 0.88708	556000
452	0.88413	+ OR - 0.00098	0.88315 TO 0.88511	0.88217 TO 0.88608	0.88120 TO 0.88706	551000
457	0.88423	+ OR - 0.00098	0.88325 TO 0.88522	0.88227 TO 0.88620	0.88129 TO 0.88718	546000
462	0.88451	+ OR - 0.00098	0.88353 TO 0.88549	0.88255 TO 0.88647	0.88157 TO 0.88745	541000
467	0.88457	+ OR - 0.00099	0.88358 TO 0.88555	0.88260 TO 0.88654	0.88161 TO 0.88752	536000
472	0.88450	+ OR - 0.00098	0.88352 TO 0.88548	0.88254 TO 0.88646	0.88156 TO 0.88744	531000
477	0.88452	+ OR - 0.00099	0.88353 TO 0.88551	0.88255 TO 0.88649	0.88156 TO 0.88748	526000
482	0.88466	+ OR - 0.00099	0.88367 TO 0.88565	0.88268 TO 0.88664	0.88169 TO 0.88763	521000
487	0.88468	+ OR - 0.00100	0.88368 TO 0.88567	0.88268 TO 0.88667	0.88169 TO 0.88766	516000
492	0.88460	+ OR - 0.00100	0.88360 TO 0.88560	0.88260 TO 0.88660	0.88160 TO 0.88760	511000
497	0.88466	+ OR - 0.00100	0.88365 TO 0.88566	0.88265 TO 0.88667	0.88165 TO 0.88767	506000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
502	0.88470	+ OR - 0.00101	0.88368 TO 0.88571	0.88267 TO 0.88672	0.88166 TO 0.88773	501000
507	0.88480	+ OR - 0.00101	0.88379 TO 0.88580	0.88279 TO 0.88681	0.88178 TO 0.88782	496000
512	0.88467	+ OR - 0.00101	0.88366 TO 0.88568	0.88265 TO 0.88669	0.88164 TO 0.88770	491000
517	0.88446	+ OR - 0.00101	0.88345 TO 0.88548	0.88243 TO 0.88649	0.88142 TO 0.88750	486000
522	0.88460	+ OR - 0.00102	0.88358 TO 0.88563	0.88256 TO 0.88665	0.88154 TO 0.88767	481000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

527	0.88482	+ OR - 0.00103	0.88379 TO 0.88585	0.88277 TO 0.88687	0.88174 TO 0.88790	476000
532	0.88509	+ OR - 0.00102	0.88407 TO 0.88611	0.88305 TO 0.88713	0.88203 TO 0.88815	471000
537	0.88521	+ OR - 0.00102	0.88419 TO 0.88623	0.88316 TO 0.88725	0.88214 TO 0.88828	466000
542	0.88513	+ OR - 0.00103	0.88410 TO 0.88616	0.88307 TO 0.88719	0.88204 TO 0.88822	461000
547	0.88507	+ OR - 0.00104	0.88403 TO 0.88611	0.88299 TO 0.88715	0.88194 TO 0.88819	456000
552	0.88500	+ OR - 0.00104	0.88396 TO 0.88604	0.88292 TO 0.88708	0.88188 TO 0.88812	451000
557	0.88505	+ OR - 0.00105	0.88400 TO 0.88610	0.88295 TO 0.88715	0.88191 TO 0.88819	446000
562	0.88490	+ OR - 0.00104	0.88385 TO 0.88594	0.88281 TO 0.88698	0.88177 TO 0.88802	441000
567	0.88493	+ OR - 0.00105	0.88388 TO 0.88597	0.88283 TO 0.88702	0.88178 TO 0.88807	436000
572	0.88505	+ OR - 0.00106	0.88399 TO 0.88610	0.88294 TO 0.88716	0.88188 TO 0.88821	431000
577	0.88493	+ OR - 0.00107	0.88386 TO 0.88599	0.88280 TO 0.88706	0.88173 TO 0.88813	426000
582	0.88485	+ OR - 0.00108	0.88378 TO 0.88593	0.88270 TO 0.88700	0.88162 TO 0.88808	421000
587	0.88489	+ OR - 0.00109	0.88380 TO 0.88598	0.88272 TO 0.88706	0.88163 TO 0.88815	416000
592	0.88487	+ OR - 0.00109	0.88378 TO 0.88597	0.88268 TO 0.88706	0.88159 TO 0.88815	411000
597	0.88492	+ OR - 0.00110	0.88382 TO 0.88601	0.88273 TO 0.88711	0.88163 TO 0.88820	406000
602	0.88504	+ OR - 0.00111	0.88393 TO 0.88614	0.88283 TO 0.88725	0.88172 TO 0.88836	401000
607	0.88517	+ OR - 0.00111	0.88406 TO 0.88629	0.88295 TO 0.88740	0.88184 TO 0.88851	396000
612	0.88533	+ OR - 0.00112	0.88421 TO 0.88645	0.88309 TO 0.88757	0.88197 TO 0.88869	391000
617	0.88542	+ OR - 0.00113	0.88429 TO 0.88654	0.88316 TO 0.88767	0.88203 TO 0.88880	386000
622	0.88547	+ OR - 0.00113	0.88433 TO 0.88660	0.88320 TO 0.88774	0.88206 TO 0.88887	381000
627	0.88560	+ OR - 0.00114	0.88446 TO 0.88675	0.88331 TO 0.88789	0.88217 TO 0.88903	376000
632	0.88574	+ OR - 0.00115	0.88459 TO 0.88689	0.88344 TO 0.88804	0.88229 TO 0.88919	371000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
637	0.88581	+ OR - 0.00116	0.88465 TO 0.88697	0.88348 TO 0.88814	0.88232 TO 0.88930	366000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

642	0.88592	+ OR - 0.00117	0.88475 TO 0.88708	0.88359 TO 0.88825	0.88242 TO 0.88941	361000
647	0.88595	+ OR - 0.00118	0.88477 TO 0.88712	0.88360 TO 0.88830	0.88242 TO 0.88947	356000
652	0.88599	+ OR - 0.00119	0.88480 TO 0.88717	0.88361 TO 0.88836	0.88243 TO 0.88955	351000
657	0.88615	+ OR - 0.00120	0.88495 TO 0.88735	0.88375 TO 0.88854	0.88255 TO 0.88974	346000
662	0.88628	+ OR - 0.00121	0.88507 TO 0.88749	0.88386 TO 0.88870	0.88265 TO 0.88991	341000
667	0.88620	+ OR - 0.00122	0.88498 TO 0.88742	0.88376 TO 0.88864	0.88254 TO 0.88987	336000
672	0.88598	+ OR - 0.00123	0.88474 TO 0.88721	0.88351 TO 0.88844	0.88228 TO 0.88967	331000
677	0.88627	+ OR - 0.00124	0.88503 TO 0.88751	0.88380 TO 0.88874	0.88256 TO 0.88998	326000
682	0.88595	+ OR - 0.00123	0.88472 TO 0.88718	0.88349 TO 0.88842	0.88225 TO 0.88965	321000
687	0.88572	+ OR - 0.00124	0.88448 TO 0.88696	0.88324 TO 0.88820	0.88200 TO 0.88943	316000
692	0.88580	+ OR - 0.00124	0.88455 TO 0.88704	0.88331 TO 0.88828	0.88207 TO 0.88953	311000
697	0.88607	+ OR - 0.00126	0.88481 TO 0.88732	0.88356 TO 0.88858	0.88230 TO 0.88984	306000
702	0.88586	+ OR - 0.00127	0.88458 TO 0.88713	0.88331 TO 0.88840	0.88204 TO 0.88967	301000
707	0.88601	+ OR - 0.00128	0.88473 TO 0.88728	0.88345 TO 0.88856	0.88218 TO 0.88984	296000
712	0.88615	+ OR - 0.00129	0.88486 TO 0.88744	0.88357 TO 0.88873	0.88228 TO 0.89002	291000
717	0.88617	+ OR - 0.00131	0.88486 TO 0.88748	0.88355 TO 0.88879	0.88225 TO 0.89010	286000
722	0.88619	+ OR - 0.00132	0.88486 TO 0.88751	0.88354 TO 0.88884	0.88221 TO 0.89016	281000
727	0.88625	+ OR - 0.00134	0.88491 TO 0.88759	0.88356 TO 0.88894	0.88222 TO 0.89028	276000
732	0.88644	+ OR - 0.00135	0.88509 TO 0.88779	0.88374 TO 0.88914	0.88239 TO 0.89049	271000
737	0.88629	+ OR - 0.00134	0.88495 TO 0.88764	0.88361 TO 0.88898	0.88226 TO 0.89032	266000
742	0.88619	+ OR - 0.00137	0.88482 TO 0.88755	0.88346 TO 0.88892	0.88209 TO 0.89028	261000
747	0.88619	+ OR - 0.00139	0.88480 TO 0.88758	0.88342 TO 0.88897	0.88203 TO 0.89036	256000
752	0.88613	+ OR - 0.00141	0.88472 TO 0.88754	0.88331 TO 0.88895	0.88190 TO 0.89036	251000
757	0.88601	+ OR - 0.00143	0.88458 TO 0.88744	0.88314 TO 0.88888	0.88171 TO 0.89031	246000
762	0.88621	+ OR - 0.00145	0.88477 TO 0.88766	0.88332 TO 0.88910	0.88187 TO 0.89055	241000
767	0.88626	+ OR - 0.00147	0.88480 TO 0.88773	0.88333 TO 0.88920	0.88186 TO 0.89067	236000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)							
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES	
772	0.88606	+ OR - 0.00149	0.88457 TO 0.88754	0.88309 TO 0.88903	0.88160 TO 0.89052	231000	
777	0.88607	+ OR - 0.00151	0.88456 TO 0.88757	0.88305 TO 0.88908	0.88155 TO 0.89059	226000	
782	0.88575	+ OR - 0.00151	0.88424 TO 0.88726	0.88273 TO 0.88877	0.88121 TO 0.89028	221000	
787	0.88527	+ OR - 0.00151	0.88376 TO 0.88677	0.88225 TO 0.88828	0.88074 TO 0.88979	216000	
792	0.88509	+ OR - 0.00152	0.88357 TO 0.88661	0.88204 TO 0.88813	0.88052 TO 0.88965	211000	
797	0.88572	+ OR - 0.00152	0.88420 TO 0.88724	0.88268 TO 0.88877	0.88116 TO 0.89029	206000	
802	0.88602	+ OR - 0.00155	0.88447 TO 0.88757	0.88293 TO 0.88911	0.88138 TO 0.89066	201000	
807	0.88647	+ OR - 0.00157	0.88490 TO 0.88804	0.88333 TO 0.88961	0.88176 TO 0.89118	196000	
812	0.88685	+ OR - 0.00159	0.88526 TO 0.88844	0.88367 TO 0.89004	0.88207 TO 0.89163	191000	
817	0.88676	+ OR - 0.00163	0.88513 TO 0.88839	0.88351 TO 0.89001	0.88188 TO 0.89164	186000	
822	0.88626	+ OR - 0.00165	0.88462 TO 0.88791	0.88297 TO 0.88955	0.88133 TO 0.89120	181000	
827	0.88660	+ OR - 0.00168	0.88492 TO 0.88828	0.88324 TO 0.88996	0.88156 TO 0.89165	176000	
832	0.88714	+ OR - 0.00170	0.88544 TO 0.88883	0.88374 TO 0.89053	0.88205 TO 0.89223	171000	
837	0.88706	+ OR - 0.00172	0.88534 TO 0.88878	0.88362 TO 0.89051	0.88190 TO 0.89223	166000	
842	0.88724	+ OR - 0.00177	0.88547 TO 0.88901	0.88370 TO 0.89078	0.88193 TO 0.89255	161000	
847	0.88714	+ OR - 0.00182	0.88532 TO 0.88895	0.88351 TO 0.89077	0.88169 TO 0.89258	156000	
852	0.88731	+ OR - 0.00184	0.88547 TO 0.88914	0.88364 TO 0.89098	0.88180 TO 0.89281	151000	
857	0.88651	+ OR - 0.00181	0.88470 TO 0.88832	0.88289 TO 0.89012	0.88109 TO 0.89193	146000	
862	0.88609	+ OR - 0.00180	0.88428 TO 0.88789	0.88248 TO 0.88970	0.88068 TO 0.89150	141000	
867	0.88607	+ OR - 0.00185	0.88422 TO 0.88792	0.88237 TO 0.88977	0.88052 TO 0.89162	136000	
872	0.88572	+ OR - 0.00187	0.88385 TO 0.88759	0.88198 TO 0.88946	0.88011 TO 0.89133	131000	
877	0.88594	+ OR - 0.00192	0.88402 TO 0.88785	0.88211 TO 0.88977	0.88019 TO 0.89168	126000	
882	0.88676	+ OR - 0.00193	0.88482 TO 0.88869	0.88289 TO 0.89063	0.88095 TO 0.89256	121000	

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

887	0.88655	+ OR - 0.00201	0.88453 TO 0.88856	0.88252 TO 0.89057	0.88051 TO 0.89258	116000
892	0.88648	+ OR - 0.00207	0.88442 TO 0.88855	0.88235 TO 0.89062	0.88029 TO 0.89268	111000
897	0.88761	+ OR - 0.00206	0.88555 TO 0.88968	0.88349 TO 0.89174	0.88142 TO 0.89380	106000
902	0.88727	+ OR - 0.00210	0.88517 TO 0.88937	0.88307 TO 0.89147	0.88097 TO 0.89357	101000
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)						
NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
907	0.88715	+ OR - 0.00219	0.88496 TO 0.88934	0.88277 TO 0.89152	0.88058 TO 0.89371	96000
912	0.88806	+ OR - 0.00224	0.88582 TO 0.89030	0.88358 TO 0.89254	0.88134 TO 0.89478	91000
917	0.88840	+ OR - 0.00235	0.88605 TO 0.89075	0.88370 TO 0.89311	0.88135 TO 0.89546	86000
922	0.88737	+ OR - 0.00241	0.88496 TO 0.88978	0.88255 TO 0.89218	0.88014 TO 0.89459	81000
927	0.88714	+ OR - 0.00247	0.88467 TO 0.88961	0.88220 TO 0.89208	0.87973 TO 0.89455	76000
932	0.88723	+ OR - 0.00258	0.88465 TO 0.88981	0.88207 TO 0.89239	0.87949 TO 0.89497	71000
937	0.88755	+ OR - 0.00266	0.88489 TO 0.89021	0.88223 TO 0.89286	0.87957 TO 0.89552	66000
942	0.88802	+ OR - 0.00257	0.88545 TO 0.89059	0.88287 TO 0.89316	0.88030 TO 0.89573	61000
947	0.88793	+ OR - 0.00273	0.88520 TO 0.89066	0.88247 TO 0.89339	0.87974 TO 0.89612	56000
952	0.88768	+ OR - 0.00295	0.88473 TO 0.89063	0.88178 TO 0.89358	0.87883 TO 0.89653	51000
957	0.88706	+ OR - 0.00320	0.88386 TO 0.89026	0.88066 TO 0.89346	0.87746 TO 0.89666	46000
962	0.88575	+ OR - 0.00329	0.88246 TO 0.88903	0.87918 TO 0.89232	0.87589 TO 0.89560	41000
967	0.88708	+ OR - 0.00359	0.88349 TO 0.89067	0.87991 TO 0.89425	0.87632 TO 0.89784	36000
972	0.88920	+ OR - 0.00391	0.88529 TO 0.89311	0.88138 TO 0.89702	0.87747 TO 0.90094	31000
977	0.88785	+ OR - 0.00379	0.88406 TO 0.89164	0.88027 TO 0.89543	0.87648 TO 0.89922	26000
982	0.88718	+ OR - 0.00439	0.88279 TO 0.89157	0.87841 TO 0.89595	0.87402 TO 0.90034	21000
987	0.88874	+ OR - 0.00462	0.88413 TO 0.89336	0.87951 TO 0.89798	0.87489 TO 0.90260	16000
992	0.88754	+ OR - 0.00641	0.88114 TO 0.89395	0.87473 TO 0.90036	0.86832 TO 0.90676	11000
997	0.88286	+ OR - 0.00981	0.87305 TO 0.89266	0.86324 TO 0.90247	0.85344 TO 0.91228	6000

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

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RANDOM NUMBER= 4BC7185F6126
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
FREQUENCY FOR GENERATIONS 4 TO 1003

0.7895 TO 0.7925 *
0.7925 TO 0.7954
0.7954 TO 0.7984
0.7984 TO 0.8014
0.8014 TO 0.8043
0.8043 TO 0.8073
0.8073 TO 0.8103
0.8103 TO 0.8132 *
0.8132 TO 0.8162
0.8162 TO 0.8192 *
0.8192 TO 0.8221
0.8221 TO 0.8251 **
0.8251 TO 0.8281 ***
0.8281 TO 0.8310 **
0.8310 TO 0.8340 *****
0.8340 TO 0.8369 *****
0.8369 TO 0.8399 *****
0.8399 TO 0.8429 *****
0.8429 TO 0.8458 *****
0.8458 TO 0.8488 *****
0.8488 TO 0.8518 *****
0.8518 TO 0.8547 *****
0.8547 TO 0.8577 *****
0.8577 TO 0.8607 *****
0.8607 TO 0.8636 *****
0.8636 TO 0.8666 *****
0.8666 TO 0.8695 *****
0.8695 TO 0.8725 *****
0.8725 TO 0.8755 *****
0.8755 TO 0.8784 *****
0.8784 TO 0.8814 *****
0.8814 TO 0.8844 *****
0.8844 TO 0.8873 *****
0.8873 TO 0.8903 *****
0.8903 TO 0.8933 *****
0.8933 TO 0.8962 *****
0.8962 TO 0.8992 *****
0.8992 TO 0.9022 *****
0.9022 TO 0.9051 *****
0.9051 TO 0.9081 *****
0.9081 TO 0.9110 *****
0.9110 TO 0.9140 *****
0.9140 TO 0.9170 *****
0.9170 TO 0.9199 *****
0.9199 TO 0.9229 *****
0.9229 TO 0.9259 *****
0.9259 TO 0.9288 *****
0.9288 TO 0.9318 *****
0.9318 TO 0.9348 ****
0.9348 TO 0.9377 ***
0.9377 TO 0.9407 ****
0.9407 TO 0.9437 ****
0.9437 TO 0.9466 ***
0.9466 TO 0.9496 ***
0.9496 TO 0.9525 *
0.9525 TO 0.9555
0.9555 TO 0.9585 **
0.9585 TO 0.9614
0.9614 TO 0.9644 **
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
FREQUENCY FOR GENERATIONS 254 TO 1003

0.7895 TO 0.7925	
0.7925 TO 0.7954	
0.7954 TO 0.7984	
0.7984 TO 0.8014	
0.8014 TO 0.8043	
0.8043 TO 0.8073	
0.8073 TO 0.8103	
0.8103 TO 0.8132	*
0.8132 TO 0.8162	
0.8162 TO 0.8192	*
0.8192 TO 0.8221	
0.8221 TO 0.8251	**
0.8251 TO 0.8281	**
0.8281 TO 0.8310	*
0.8310 TO 0.8340	***
0.8340 TO 0.8369	***
0.8369 TO 0.8399	*****
0.8399 TO 0.8429	*****
0.8429 TO 0.8458	*****
0.8458 TO 0.8488	*****
0.8488 TO 0.8518	*****
0.8518 TO 0.8547	*****
0.8547 TO 0.8577	*****
0.8577 TO 0.8607	*****
0.8607 TO 0.8636	*****
0.8636 TO 0.8666	*****
0.8666 TO 0.8695	*****
0.8695 TO 0.8725	*****
0.8725 TO 0.8755	*****
0.8755 TO 0.8784	*****
0.8784 TO 0.8814	*****
0.8814 TO 0.8844	*****
0.8844 TO 0.8873	*****
0.8873 TO 0.8903	*****
0.8903 TO 0.8933	*****
0.8933 TO 0.8962	*****
0.8962 TO 0.8992	*****
0.8992 TO 0.9022	*****
0.9022 TO 0.9051	*****
0.9051 TO 0.9081	*****
0.9081 TO 0.9110	*****
0.9110 TO 0.9140	*****
0.9140 TO 0.9170	*****
0.9170 TO 0.9199	*****
0.9199 TO 0.9229	*****
0.9229 TO 0.9259	*****
0.9259 TO 0.9288	***
0.9288 TO 0.9318	*****
0.9318 TO 0.9348	***
0.9348 TO 0.9377	*
0.9377 TO 0.9407	***
0.9407 TO 0.9437	***
0.9437 TO 0.9466	
0.9466 TO 0.9496	***
0.9496 TO 0.9525	*
0.9525 TO 0.9555	
0.9555 TO 0.9585	
0.9585 TO 0.9614	
0.9614 TO 0.9644	**

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
FREQUENCY FOR GENERATIONS 504 TO 1003

```
0.7895 TO 0.7925
0.7925 TO 0.7954
0.7954 TO 0.7984
0.7984 TO 0.8014
0.8014 TO 0.8043
0.8043 TO 0.8073
0.8073 TO 0.8103
0.8103 TO 0.8132 *
0.8132 TO 0.8162
0.8162 TO 0.8192 *
0.8192 TO 0.8221
0.8221 TO 0.8251 *
0.8251 TO 0.8281 *
0.8281 TO 0.8310
0.8310 TO 0.8340
0.8340 TO 0.8369 ***
0.8369 TO 0.8399 *****
0.8399 TO 0.8429 ****
0.8429 TO 0.8458 *****
0.8458 TO 0.8488 *****
0.8488 TO 0.8518 *****
0.8518 TO 0.8547 *****
0.8547 TO 0.8577 *****
0.8577 TO 0.8607 *****
0.8607 TO 0.8636 *****
0.8636 TO 0.8666 *****
0.8666 TO 0.8695 *****
0.8695 TO 0.8725 *****
0.8725 TO 0.8755 *****
0.8755 TO 0.8784 *****
0.8784 TO 0.8814 *****
0.8814 TO 0.8844 *****
0.8844 TO 0.8873 *****
0.8873 TO 0.8903 *****
0.8903 TO 0.8933 *****
0.8933 TO 0.8962 *****
0.8962 TO 0.8992 *****
0.8992 TO 0.9022 *****
0.9022 TO 0.9051 *****
0.9051 TO 0.9081 *****
0.9081 TO 0.9110 *****
0.9110 TO 0.9140 *****
0.9140 TO 0.9170 *****
0.9170 TO 0.9199 ***
0.9199 TO 0.9229 *****
0.9229 TO 0.9259 *****
0.9259 TO 0.9288 **
0.9288 TO 0.9318 *****
0.9318 TO 0.9348 ***
0.9348 TO 0.9377
0.9377 TO 0.9407 *
0.9407 TO 0.9437 **
0.9437 TO 0.9466
0.9466 TO 0.9496 *
0.9496 TO 0.9525 *
0.9525 TO 0.9555
0.9555 TO 0.9585
0.9585 TO 0.9614
0.9614 TO 0.9644 *
```

Figure 6.7-6 CSAS25 Output for Canistered Yankee-Class Fuel - Accident Conditions
(Continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
FREQUENCY FOR GENERATIONS 754 TO 1003

0.7895 TO 0.7925	
0.7925 TO 0.7954	
0.7954 TO 0.7984	
0.7984 TO 0.8014	
0.8014 TO 0.8043	
0.8043 TO 0.8073	
0.8073 TO 0.8103	
0.8103 TO 0.8132	
0.8132 TO 0.8162	
0.8162 TO 0.8192	
0.8192 TO 0.8221	
0.8221 TO 0.8251	
0.8251 TO 0.8281	
0.8281 TO 0.8310	
0.8310 TO 0.8340	
0.8340 TO 0.8369	***
0.8369 TO 0.8399	**
0.8399 TO 0.8429	**
0.8429 TO 0.8458	**
0.8458 TO 0.8488	**
0.8488 TO 0.8518	*****
0.8518 TO 0.8547	*
0.8547 TO 0.8577	****
0.8577 TO 0.8607	*****
0.8607 TO 0.8636	*****
0.8636 TO 0.8666	*****
0.8666 TO 0.8695	*****
0.8695 TO 0.8725	*****
0.8725 TO 0.8755	*****
0.8755 TO 0.8784	*****
0.8784 TO 0.8814	*****
0.8814 TO 0.8844	*****
0.8844 TO 0.8873	*****
0.8873 TO 0.8903	*****
0.8903 TO 0.8933	*****
0.8933 TO 0.8962	*****
0.8962 TO 0.8992	*****
0.8992 TO 0.9022	*****
0.9022 TO 0.9051	*****
0.9051 TO 0.9081	*****
0.9081 TO 0.9110	*****
0.9110 TO 0.9140	*****
0.9140 TO 0.9170	*****
0.9170 TO 0.9199	*
0.9199 TO 0.9229	**
0.9229 TO 0.9259	****
0.9259 TO 0.9288	
0.9288 TO 0.9318	*****
0.9318 TO 0.9348	**
0.9348 TO 0.9377	
0.9377 TO 0.9407	
0.9407 TO 0.9437	*
0.9437 TO 0.9466	
0.9466 TO 0.9496	
0.9496 TO 0.9525	*
0.9525 TO 0.9555	
0.9555 TO 0.9585	
0.9585 TO 0.9614	
0.9614 TO 0.9644	*

Figure 6.7-7 CSAS25 Input/Output for Reconfigured Fuel Assembly Analysis

```
PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
Failed Fuel Assembly Analysis - UNC Type fuel - Zr/SS clads homogenized

UNITED NUCLEAR ASSEMBLY

MAXIMUM INITAIL ENRICHMENT = 4.0% U-235
CLAD MATERIAL              = ZIRCAOLLY

ANALYSIS PREFORMED FOR THE YANKEE ROWE
SPENT FUEL STORAGE PROJECT

27GROUPNDF4 LATTICECELL
UO2      1  0.95          293.0 92235 4.0 92238 96.0 END
ZIRCALLOY 2  1.0          293.0 END
H2O      3  1.0          293.0 END
AL       4  1.0          293.0 END
SS304    5  1.0          293.0 END
B-10     6  DEN=2.6266 0.0450 293.0 END
B-11     6  DEN=2.6266 0.2731 293.0 END
C        6  DEN=2.6266 0.0925 293.0 END
AL       6  DEN=2.6266 0.5744 293.0 END
H2O      7  1.0          293.0 END
ZIRCALLOY 8  DEN=4.6640 0.3080 293.0 END
H2O      8  DEN=4.6640 0.0917 293.0 END
SS304    8  DEN=4.6640 0.6003 293.0 END
END COMP
SQUAREPITCH 1.905 0.7887 1 3 1.27 8 0.8052 7 END
Failed Fuel Assembly Analysis - UNC Type fuel - Zr/SS clads homogenized
READ PARAM RUN=yes PLT=no GEN=1003 NPG=1000 TME=500 END PARAM
READ GEOM

WATER LEVEL UNIT CELLS

UNIT 1
COM='FUEL PIN CELL - BETWEEN DISKS'
CYLINDER 1 1 0.3943 2P2.1654
CYLINDER 7 1 0.4026 2P2.1654
CYLINDER 8 1 0.635 2P2.1654
CUBOID 3 1 4P0.9525 2P2.1654

DISK LEVEL UNIT CELLS (BOTH SS AND AL)

UNIT 2
COM='FUEL PIN CELL - WITH SS/AL DISK'
CYLINDER 1 1 0.3943 2P0.635
CYLINDER 7 1 0.4026 2P0.635
CYLINDER 8 1 0.635 2P0.635
CUBOID 3 1 4P0.9525 2P0.635

WATER LEVEL BORAL SHEETS

UNIT 14
COM='X-X BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P9.144 2P0.0318 2P2.1654
CUBOID 4 1 2P9.144 2P0.0953 2P2.1654
UNIT 15
COM='Y-Y BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P0.0318 2P9.144 2P2.1654
CUBOID 4 1 2P0.0953 2P9.144 2P2.1654

DISK LEVEL BORAL SHEETS (AL AND SS)
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
UNIT 16
COM='X-X BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P9.144 2P0.0318 2P0.635
CUBOID 4 1 2P9.144 2P0.0953 2P0.635
UNIT 17
COM='Y-Y BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P0.0318 2P9.144 2P0.635
CUBOID 4 1 2P0.0953 2P9.144 2P0.635
'
ASSEMBLY ARRAYS
'
UNIT 18
COM='WATER LEVEL FUEL TUBE'
ARRAY 1 -7.6200 -7.6200 -2.1654
CUBOID 3 1 4P9.04875 2P2.1654
CUBOID 5 1 4P9.37387 2P2.1654
CUBOID 3 1 4P9.906 2P2.1654
CUBOID 5 1 4P10.028 2P2.1654
CUBOID 3 1 4P10.2187 2P2.1654
HOLE 14 0.0 10.1234 0.0
HOLE 14 0.0 -10.1234 0.0
HOLE 15 10.1234 0.0 0.0
HOLE 15 -10.1234 0.0 0.0
CUBOID 5 1 4P10.267 2P2.1654
CUBOID 3 1 4P10.4826 2P2.1654
CUBOID 3 1 4P11.594 2P2.1654
'
UNIT 19
COM='SS DISK LEVEL FUEL TUBE'
ARRAY 2 -7.6200 -7.6200 -0.635
CUBOID 3 1 4P9.04875 2P0.635
CUBOID 5 1 4P9.37387 2P0.635
CUBOID 3 1 4P9.906 2P0.635
CUBOID 5 1 4P10.028 2P0.635
CUBOID 3 1 4P10.2187 2P0.635
HOLE 16 0.0 10.1234 0.0
HOLE 16 0.0 -10.1234 0.0
HOLE 17 10.1234 0.0 0.0
HOLE -17 -10.1234 0.0 0.0
CUBOID 5 1 4P10.267 2P0.635
CUBOID 3 1 4P10.4826 2P0.635
CUBOID 5 1 4P11.594 2P0.635
'
UNIT 20
COM='AL DISK LEVEL FUEL TUBE'
ARRAY 2 -7.6200 -7.6200 -0.635
CUBOID 3 1 4P9.04875 2P0.635
CUBOID 5 1 4P9.37387 2P0.635
CUBOID 3 1 4P9.906 2P0.635
CUBOID 5 1 4P10.028 2P0.635
CUBOID 3 1 4P10.2187 2P0.635
HOLE 16 0.0 10.1234 0.0
HOLE 16 0.0 -10.1234 0.0
HOLE 17 10.1234 0.0 0.0
HOLE 17 -10.1234 0.0 0.0
CUBOID 5 1 4P10.267 2P0.635
CUBOID 3 1 4P10.4445 2P0.635
CUBOID 4 1 4P11.594 2P0.635
'
GLOBAL UNIT
'
GLOBAL UNIT 40
ARRAY 30 -11.594 -11.594 -5.6008
END GEOM

READ ARRAY
ARA=1 NUX=8 NUY=8 NUZ=1 FILL
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
1 1 1 1 1 1 1 1
END FILL
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
ARA=2 NUX=8 NUY=8 NUZ=1 FILL
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
2 2 2 2 2 2 2 2
END FILL

ARA=30 NUX=1 NUY=1 NUZ=4
FILL
20
18
19
18
END FILL
END ARRAY

READ BOUNDS ZFC=PER YXF=REFLECT END BOUNDS

READ PLOT
SCR=YES PIC=MAT LPI=10
TTL='BASKET X-Y CROSS SECTION AT Z= -2.1654 WATER LEVEL'
XUL= -12 YUL= 12 ZUL= -2.1654
XLR= 12 YLR= -12 ZLR= -2.1654
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= -4.9657 AL DISK LEVEL'
XUL= -12 YUL= 12 ZUL= -4.9657
XLR= 12 YLR= -12 ZLR= -4.9657
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 0.25 DISK LEVEL'
XUL= -12 YUL= 12 ZUL= 0.25
XLR= 12 YLR= -12 ZLR= 0.25
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION AT Y=0.9525 (MIDDLE OF FUEL PIN)'
XUL= -11.6 YUL=0.9525 ZUL= 5.608
XLR= 11.6 YLR=0.9525 ZLR=-5.608
UAX=1.0 WDN=-1.0 NAX=1500 END
END PLOT
END DATA
```

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
-----  
*****  
*****  
***** PROGRAM: CSAS *****  
*****  
***** CREATION DATE: 03-08-96 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
***** PRODUCTION CODE: CSAS *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 04/03/97 *****  
*****  
***** TIME OF EXECUTION: 20:59:48 *****  
*****  
-----
```

**** PROBLEM PARAMETERS ****

***** PROBLEM COMPOSITION DESCRIPTION *****

SC	ZIRCALLOY	STANDARD COMPOSITION
MX	2	MIXTURE NO.
VF	1.0000	VOLUME FRACTION
ROTH	6.5600	THEORETICAL DENSITY
NEL	1	NO. ELEMENTS
ICP	1	0/1 MIXTURE/COMPOUND
TEMP	293.0	DEG KELVIN

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
40302      1.00 ATOM/MOLECULE
END

SC H2O      STANDARD COMPOSITION
MX          3 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        0.9982 THEORETICAL DENSITY
NEL         2 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           1001      2.00 ATOMS/MOLECULE
           8016      1.00 ATOM/MOLECULE
END

SC AL        STANDARD COMPOSITION
MX          4 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        2.7020 THEORETICAL DENSITY
NEL         1 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           13027     1.00 ATOM/MOLECULE
END

SC SS304     STANDARD COMPOSITION
MX          5 MIXTURE NO.
VF          1.0000 VOLUME FRACTION
ROTH        7.9200 THEORETICAL DENSITY
NEL         4 NO. ELEMENTS
ICP         0 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           24304     19.000 WT%
           25055     2.000 WT%
           26304     69.500 WT%
           28304     9.500 WT%
END

SC B-10      STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF          0.0450 VOLUME FRACTION
ROTH        2.6266 SPECIFIED DENSITY
NEL         1 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           5010      1.00 ATOM/MOLECULE
END

SC B-11      STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF          0.2731 VOLUME FRACTION
ROTH        2.6266 SPECIFIED DENSITY
NEL         1 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           5011      1.00 ATOM/MOLECULE
END

SC C         STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF          0.0925 VOLUME FRACTION
ROTH        2.6266 SPECIFIED DENSITY
NEL         1 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           6012      1.00 ATOM/MOLECULE
END

SC AL        STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF          0.5744 VOLUME FRACTION
ROTH        2.6266 SPECIFIED DENSITY
NEL         1 NO. ELEMENTS
ICP         1 0/1 MIXTURE/COMPOUND
TEMP        293.0 DEG KELVIN
           13027     1.00 ATOM/MOLECULE
END
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
SC H2O          STANDARD COMPOSITION
MX              7 MIXTURE NO.
VF              1.0000 VOLUME FRACTION
ROTH            0.9982 THEORETICAL DENSITY
NEL             2 NO. ELEMENTS
ICP             1 0/1 MIXTURE/COMPOUND
TEMP            293.0 DEG KELVIN
                1001      2.00 ATOMS/MOLECULE
                8016      1.00 ATOM/MOLECULE
END

SC ZIRCALLOY    STANDARD COMPOSITION
MX              8 MIXTURE NO.
VF              0.3080 VOLUME FRACTION
ROTH            4.6640 SPECIFIED DENSITY
NEL             1 NO. ELEMENTS
ICP             1 0/1 MIXTURE/COMPOUND
TEMP            293.0 DEG KELVIN
                40302     1.00 ATOM/MOLECULE
END

SC H2O          STANDARD COMPOSITION
MX              8 MIXTURE NO.
VF              0.0917 VOLUME FRACTION
ROTH            4.6640 SPECIFIED DENSITY
NEL             2 NO. ELEMENTS
ICP             1 0/1 MIXTURE/COMPOUND
TEMP            293.0 DEG KELVIN
                1001      2.00 ATOMS/MOLECULE
                8016      1.00 ATOM/MOLECULE
END

SC SS304        STANDARD COMPOSITION
MX              8 MIXTURE NO.
VF              0.6003 VOLUME FRACTION
ROTH            4.6640 SPECIFIED DENSITY
NEL             4 NO. ELEMENTS
ICP             0 0/1 MIXTURE/COMPOUND
TEMP            293.0 DEG KELVIN
                24304     19.000 WT%
                25055     2.000 WT%
                26304     69.500 WT%
                28304     9.500 WT%
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH           1.9050 CM CENTER TO CENTER SPACING
FUELOD          0.7887 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL           1 MIXTURE NO. OF FUEL
MMOD            3 MIXTURE NO. OF MODERATOR
CLADOD          1.2700 CM CLAD OUTER DIAMETER
MCLAD           8 MIXTURE NO. OF CLAD
GAPOD           0.8052 CM GAP OUTER DIAMETER
MGAP            7 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY
ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD
```

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
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*****  
***** PROGRAM: O0O002 *****  
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***** CREATION DATE: 09-28-95 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
*****  
***** PRODUCTION CODE: NITAWL *****  
*****  
***** VERSION: 3.0 *****  
*****  
***** JOBNAM E : SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 04/03/97 *****  
*****  
***** TIME OF EXECUTION: 20:59:51 *****  
*****  
*****
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
-1Q ARRAY HAS      1 ENTRIES.

0Q ARRAY HAS      9 ENTRIES.

1Q ARRAY HAS     12 ENTRIES.

SELECT 24 NUCLIDES FROM THE MASTER  LIBRARY ON LOGICAL  1
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL  2
      0 NUCLIDES FROM THE WORKING LIBRARY ON LOGICAL  3
      TO CREATE THE NEW WORKING LIBRARY ON LOGICAL  4

      6 RESONANCE CALCULATIONS HAVE BEEN REQUESTED
      -1 OUTPUT OPTION FOR AMPX FORMATTED CROSS SECTION DATA
      2001 MAXIMUM NUMBER OF RESONANCE MESH INTERVALS
      2 ORDER OF RESONANCE LEVEL PROCESSING

THE STORAGE ALLOCATED FOR THIS CASE IS    100000 WORDS

      2Q ARRAY HAS     24 ENTRIES.

      3Q ARRAY HAS     90 ENTRIES.

      4Q ARRAY HAS     24 ENTRIES.

GENERAL INFORMATION CONCERNING CROSS SECTION LIBRARY
TAPE IDENTIFICATION NUMBER      4321
NUMBER OF NUCLIDES ON TAPE      24
NUMBER OF NEUTRON ENERGY GROUPS  27
FIRST THERMAL NEUTRON ENERGY GROUP  15
NUMBER OF GAMMA ENERGY GROUPS    0

DIRECT ACCESS UNIT NUMBER  9 REQUIRES 117 BLOCKS OF LENGTH 1680 WORDS
XSDRN TAPE  4321

      SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
      BASED ON ENDF-B VERSION 4 DATA
      COMPILED FOR NRC    1/27/89
      LAST UPDATED
      L.M.PETRIE - ORNL

08/12/94

NUCLIDES FROM XSDRN TAPE
1  HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      3001001
2  HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      7001001
3  HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      8001001
4  B-10 1273 218NGP 042375 P-3 293K              UPDATED 08/12/94      6005010
5  BORON-11      ENDF/B-IV MAT 1160                UPDATED 08/12/94      6005011
6  CARBON-12     ENDF/B-IV MAT 1274/THRM1065        UPDATED 08/12/94      6006012
7  OXYGEN-16     ENDF/B-IV MAT 1276                UPDATED 08/12/94      1008016
8  OXYGEN-16     ENDF/B-IV MAT 1276                UPDATED 08/12/94      3008016
9  OXYGEN-16     ENDF/B-IV MAT 1276                UPDATED 08/12/94      7008016
10 OXYGEN-16     ENDF/B-IV MAT 1276                UPDATED 08/12/94      8008016
11 AL-27 1193 218 GP 040375(5)                   UPDATED 08/12/94      4013027
12 AL-27 1193 218 GP 040375(5)                   UPDATED 08/12/94      6013027
13 CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      5024304
14 CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      8024304
15 MANGANESE-55  ENDF/B-IV MAT 1197                UPDATED 08/12/94      5025055
16 MANGANESE-55  ENDF/B-IV MAT 1197                UPDATED 08/12/94      8025055
17 FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      5026304
18 FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      8026304
19 NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      5028304
20 NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94      8028304
21 ZIRCALLOY     ENDF/B-IV MAT 1284                UPDATED 08/12/94      2040302
22 ZIRCALLOY     ENDF/B-IV MAT 1284                UPDATED 08/12/94      8040302
23 URANIUM-235   ENDF/B-IV MAT 1261                UPDATED 08/12/94      1092235
24 URANIUM-238   ENDF/B-IV MAT 1262                UPDATED 08/12/94      1092238

HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      3001001      TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      7001001      TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

HYDROGEN      ENDF/B-IV MAT 1269/THRM1002      UPDATED 08/12/94      8001001      TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
B-10 1273 218NGP 042375 P-3 293K          UPDATED 08/12/94 6005010  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

BORON-11      ENDF/B-IV MAT 1160          UPDATED 08/12/94 6005011  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

CARBON-12     ENDF/B-IV MAT 1274/THRM1065  UPDATED 08/12/94 6006012  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

OXYGEN-16     ENDF/B-IV MAT 1276          UPDATED 08/12/94 1008016  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

OXYGEN-16     ENDF/B-IV MAT 1276          UPDATED 08/12/94 3008016  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

OXYGEN-16     ENDF/B-IV MAT 1276          UPDATED 08/12/94 7008016  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

OXYGEN-16     ENDF/B-IV MAT 1276          UPDATED 08/12/94 8008016  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5)              UPDATED 08/12/94 4013027  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

AL-27 1193 218 GP 040375(5)              UPDATED 08/12/94 6013027  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94 5024304  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'  UPDATED 08/12/94 8024304  TEMPERATURE= 293.00
                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

MANGANESE-55  ENDF/B-IV MAT 1197          UPDATED 08/12/94 5025055  TEMPERATURE= 293.00

GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)      = 54.466              TEMPERATURE(KELVIN)    = 293.000
POTENTIAL SCATTER SIGMA = 2.590              LUMPED NUCLEAR DENSITY = 1.7363295E-03
SPIN FACTOR (G)      = 14.448              LUMP DIMENSION (A-BAR) = 0.0000000E+00
INNER RADIUS          = 0.0000000E+00        DANCORFF CORRECTION (C) = 0.0000000E+00

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1  = 55.845              SIGMA(PER ABSORBER ATOM)= 3.4663022E+02
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2  = 55.925              SIGMA(PER ABSORBER ATOM)= 1.2557598E+02
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP      RES ABS      RES FISS      RES SCAT
 8      -5.518788E-04  0.000000E+00  -3.944190E-01
 9      -2.797993E-03  0.000000E+00  -2.293471E+00
10      -3.291452E-01  0.000000E+00  -3.820862E+01
11      -2.680562E+00  0.000000E+00  -1.159996E+02

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION      3.33719E+00
FISSION          0.00000E+00

                                           PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

MANGANESE-55 ENDF/B-IV MAT 1197 UPDATED 08/12/94 8025055 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 54.466 TEMPERATURE (KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 2.590 LUMPED NUCLEAR DENSITY = 6.1380991E-04
SPIN FACTOR (G) = 14.448 LUMP DIMENSION (A-BAR) = 6.3499999E-01
INNER RADIUS = 4.0259999E-01 DANCORFF CORRECTION (C) = 3.2576096E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 1.008 SIGMA (PER ABSORBER ATOM) = 9.4989429E+02

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 43.937 SIGMA (PER ABSORBER ATOM) = 6.6430096E+02

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-4.632522E-05	0.000000E+00	-2.604041E-02
9	-1.038829E-04	0.000000E+00	-9.882612E-02
10	-5.790435E-02	0.000000E+00	-2.306738E+00
11	-3.794656E-01	0.000000E+00	-1.751240E+01

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION 8.52043E+00
FISSION 0.00000E+00

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 5026304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 8026304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 5028304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375) ' UPDATED 08/12/94 8028304 TEMPERATURE= 293.00
PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

ZIRCALLOY ENDF/B-IV MAT 1284 . UPDATED 08/12/94 2040302 TEMPERATURE= 293.00

GEOMETRY HAS BEEN SET TO HOMOGENEOUS AS LBAR IS 0.0000E+00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A) = 90.436 TEMPERATURE (KELVIN) = 293.000
POTENTIAL SCATTER SIGMA = 6.385 LUMPED NUCLEAR DENSITY = 4.3307818E-02
SPIN FACTOR (G) = 1.079 LUMP DIMENSION (A-BAR) = 0.0000000E+00
INNER RADIUS = 0.0000000E+00 DANCORFF CORRECTION (C) = 0.0000000E+00

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 0-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
8	-2.531564E-03	0.000000E+00	-2.069429E+00
9	-7.143981E-02	0.000000E+00	-3.266492E+00

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```
10      -7.703653E-02    0.000000E+00    -1.746459E+00
11      -1.954898E-01    0.000000E+00    -8.103043E-01

EXCESS RESONANCE INTEGRALS

      RESOLVED

ABSORPTION      1.75363E-01
FISSION         0.00000E+00

      PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

ZIRCALLOY      ENDF/B-IV MAT 1284      UPDATED 08/12/94  8040302      TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)      = 90.436      TEMPERATURE(KELVIN)      = 293.000
POTENTIAL SCATTER SIGMA      = 6.385      LUMPED NUCLEAR DENSITY      = 9.4835665E-03
SPIN FACTOR (G)      = 1.079      LUMP DIMENSION (A-BAR)      = 6.3499999E-01
INNER RADIUS      = 4.0259999E-01      DANCOFF CORRECTION (C)      = 3.2576096E-01

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1      = 1.008      SIGMA(PER ABSORBER ATOM)= 6.1480515E-01
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2      = 40.227      SIGMA(PER ABSORBER ATOM)= 3.6691963E-01
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP      RES ABS      RES FISS      RES SCAT
8      -1.628454E-04    0.000000E+00    -1.293278E-01
9      -1.405488E-02    0.000000E+00    -5.710824E-01
10     -1.309752E-02    0.000000E+00    -3.091693E-01
11     -5.092335E-02    0.000000E+00    -2.514589E-01

EXCESS RESONANCE INTEGRALS

      RESOLVED

ABSORPTION      6.45563E-01
FISSION         0.00000E+00

      PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-235      ENDF/B-IV MAT 1261      UPDATED 08/12/94  1092235      TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)      = 233.025      TEMPERATURE(KELVIN)      = 293.000
POTENTIAL SCATTER SIGMA      = 11.500      LUMPED NUCLEAR DENSITY      = 9.4064139E-04
SPIN FACTOR (G)      = 15171.100      LUMP DIMENSION (A-BAR)      = 3.9434999E-01
INNER RADIUS      = 0.0000000E+00      DANCOFF CORRECTION (C)      = 5.1593807E-02

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1      = 15.991      SIGMA(PER ABSORBER ATOM)= 1.9199110E+02
MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2      = 238.051      SIGMA(PER ABSORBER ATOM)= 2.9209552E+02
MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
12	-3.024781E+00	-1.858535E+00	-7.440431E-02
13	-9.829280E+00	-4.807918E+00	-2.158688E-01
14	-7.256217E+00	-4.316991E+00	-5.064887E-02
15	-3.783771E-04	-2.876999E-04	3.180036E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	2.04765E+02
FISSION	1.22513E+02

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

URANIUM-238 ENDF/B-IV MAT 1262

UPDATED 08/12/94 1092238 TEMPERATURE= 293.00

RESONANCE DATA FOR THIS NUCLIDE

MASS NUMBER (A)	=	236.006	TEMPERATURE (KELVIN)	=	293.000
POTENTIAL SCATTER SIGMA	=	10.599	LUMPED NUCLEAR DENSITY	=	2.2290209E-02
SPIN FACTOR (G)	=	656.527	LUMP DIMENSION (A-BAR)	=	3.9434999E-01
INNER RADIUS	=	0.0000000E+00	DANCOFF CORRECTION (C)	=	5.1593807E-02

THE ABSORBER WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-1 = 15.991 SIGMA (PER ABSORBER ATOM) = 8.1019773E+00

MODERATOR-1 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

MASS OF MODERATOR-2 = 235.044 SIGMA (PER ABSORBER ATOM) = 5.0228214E-01

MODERATOR-2 WILL BE TREATED BY THE NORDHEIM INTEGRAL METHOD.

THIS RESONANCE MATERIAL WILL BE TREATED AS A 2-DIMENSIONAL OBJECT.

VOLUME FRACTION OF LUMP IN CELL USED TO ACCOUNT FOR SPATIAL SELF-SHIELDING=1.00000

GROUP	RES ABS	RES FISS	RES SCAT
9	-4.118901E-02	0.000000E+00	-4.144779E-01
10	-1.071716E+00	-1.898310E-05	-6.558035E+00
11	-9.741570E+00	0.000000E+00	-2.679138E+01
12	-4.269218E+01	0.000000E+00	-4.964766E+01
13	-5.346131E+01	0.000000E+00	-1.756231E+01
14	-1.034559E+02	0.000000E+00	-6.027745E+00
15	-6.122580E-07	0.000000E+00	1.186307E-06

EXCESS RESONANCE INTEGRALS

RESOLVED

ABSORPTION	2.07674E+01
FISSION	5.01450E-04

PROCESS NUMBER 1007 IS AT TEMPERATURE= 293.00

THIS XSDRN WORKING TAPE WAS CREATED 04/03/97 AT 20:59:51

THE TITLE OF THE PARENT CASE IS AS FOLLOWS

SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY

BASED ON ENDF-B VERSION 4 DATA

COMPILED FOR NRC 1/27/89

TAPE ID	4321	NUMBER OF NUCLIDES	24
NUMBER OF NEUTRON GROUPS	27	NUMBER OF GAMMA GROUPS	0
FIRST THERMAL GROUP	15	LOGICAL UNIT	4

TABLE OF CONTENTS

HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID	3001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID	7001001
HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94	ID	8001001
B-10 1273 218NGP 042375 P-3 293K		UPDATED 08/12/94	ID	6005010
BORON-11	ENDF/B-IV MAT 1160	UPDATED 08/12/94	ID	6005011
CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94	ID	6006012
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID	1008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID	3008016
OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID	7008016

OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED 08/12/94	ID 8008016
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 4013027
AL-27 1193 218 GP 040375(5)		UPDATED 08/12/94	ID 6013027
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 5024304
CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 8024304
MANGANESE-55 ENDF/B-IV MAT 1197		UPDATED 08/12/94	ID 5025055
MANGANESE-55 ENDF/B-IV MAT 1197		UPDATED 08/12/94	ID 8025055
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 5026304
FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 8026304
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 5028304
NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)'		UPDATED 08/12/94	ID 8028304
ZIRCALLOY ENDF/B-IV MAT 1284		UPDATED 08/12/94	ID 2040302
ZIRCALLOY ENDF/B-IV MAT 1284		UPDATED 08/12/94	ID 8040302
URANIUM-235 ENDF/B-IV MAT 1261		UPDATED 08/12/94	ID 1092235
URANIUM-238 ENDF/B-IV MAT 1262		UPDATED 08/12/94	ID 1092238

TAPE COPY USED 0 I/O'S, AND TOOK 0.27 SECONDS

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
*****  
***** PROGRAM: O0O009 *****  
*****  
***** CREATION DATE: 03-08-96 *****  
*****  
***** VOLUME: ENG *****  
*****  
***** LIBRARY: G:\SCALE43\EXE *****  
*****  
*****  
***** PRODUCTION CODE: KENOVA *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 04/03/97 *****  
*****  
***** TIME OF EXECUTION: 21:00:04 *****  
*****  
*****
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```

***
***                               FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED
***
*****          NUMERIC PARAMETERS          *****
***
***      TME      MAXIMUM PROBLEM TIME (MIN)      500.00      ***
***
***      TBA      TIME PER GENERATION (MIN)      0.50      ***
***
***      GEN      NUMBER OF GENERATIONS      1003      ***
***
***      NPG      NUMBER PER GENERATION      1000      ***
***
***      NSK      NUMBER OF GENERATIONS TO BE SKIPPED      3      ***
***
***      BEG      BEGINNING GENERATION NUMBER      1      ***
***
***      RES      GENERATIONS BETWEEN CHECKPOINTS      0      ***
***
***      X1D      NUMBER OF EXTRA 1-D CROSS SECTIONS      1      ***
***
***      NBK      NEUTRON BANK SIZE      1025      ***
***
***      XNB      EXTRA POSITIONS IN NEUTRON BANK      0      ***
***
***      NFB      FISSION BANK SIZE      1000      ***
***
***      XFB      EXTRA POSITIONS IN FISSION BANK      0      ***
***
***      WTA      DEFAULT VALUE OF WEIGHT AVERAGE      0.5000      ***
***
***      WTH      WEIGHT HIGH FOR SPLITTING      3.0000      ***
***
***      WTL      WEIGHT LOW FOR RUSSIAN ROULETTE      0.3333      ***
***
***      RND      STARTING RANDOM NUMBER      BB827100001      ***
***
***      NB8      NUMBER OF D.A. BLOCKS ON UNIT 8      200      ***
***
***      NL8      LENGTH OF D.A. BLOCKS ON UNIT 8      512      ***
***
***      ADJ      MODE OF CALCULATION      FORWARD      ***
***
***      INPUT DATA WRITTEN ON RESTART UNIT      NO      ***
***
***      BINARY DATA INTERFACE      YES      ***
***

```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```

*****
***
***          FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED          ***
***
***** LOGICAL PARAMETERS *****
***
*** RUN  EXECUTE PROBLEM AFTER CHECKING DATA  YES      PLT  PLOT PICTURE MAP(S)              NO ***
***
*** FLX  COMPUTE FLUX                          NO      FDN  COMPUTE FISSION DENSITIES          NO ***
***
*** SMU  COMPUTE AVG UNIT SELF-MULTIPLICATION  NO      NUB  COMPUTE NU-BAR & AVG FISSION GROUP    YES ***
***
*** MKU  COMPUTE MATRIX K-EFF BY UNIT NUMBER   NO      MKP  COMPUTE MATRIX K-EFF BY UNIT LOCATION NO ***
***
*** CKU  COMPUTE COFACTOR K-EFF BY UNIT NUMBER NO      CKP  COMPUTE COFACTOR K-EFF BY UNIT LOCATION NO ***
***
*** FMU  PRINT FISSION PROD MATRIX BY UNIT NUMBER NO  FMP  PRINT FISSION PROD MATRIX BY UNIT LOCATION NO ***
***
*** MKH  COMPUTE MATRIX K-EFF BY HOLE NUMBER   NO      MKA  COMPUTE MATRIX K-EFF BY ARRAY NUMBER    NO ***
***
*** CKH  COMPUTE COFACTOR K-EFF BY HOLE NUMBER NO      CKA  COMPUTE COFACTOR K-EFF BY ARRAY NUMBER    NO ***
***
*** FMH  PRINT FISSION PROD MATRIX BY HOLE NUMBER NO  FMA  PRINT FISSION PROD MATRIX BY ARRAY NUMBER    NO ***
***
*** HHL  COLLECT MATRIX BY HIGHEST HOLE LEVEL  NO      HAL  COLLECT MATRIX BY HIGHEST ARRAY LEVEL    NO ***
***
*** AMX  PRINT ALL MIXED CROSS SECTIONS        NO      FAR  PRINT FIS. AND ABS. BY REGION          NO ***
***
*** XS1  PRINT 1-D MIXTURE X-SECTIONS          NO      GAS  PRINT FAR BY GROUP                      NO ***
***
*** XS2  PRINT 2-D MIXTURE X-SECTIONS          NO      PAX  PRINT XSEC-ALBEDO CORRELATION TABLES NO ***
***
*** XAP  PRINT MIXTURE ANGLES & PROBABILITIES  NO      PWT  PRINT WEIGHT AVERAGE ARRAY            NO ***
***
*** PKI  PRINT FISSION SPECTRUM                NO      PGM  PRINT INPUT GEOMETRY                  NO ***
***
*** PID  PRINT EXTRA 1-D CROSS SECTIONS       NO      BUG  PRINT DEBUG INFORMATION              NO ***
***
***                                           TRK  PRINT TRACKING INFORMATION            NO ***
***
*****
PARAMETER INPUT COMPLETED

```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

```

*****
***                                     ***
***      FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED      ***
***                                     ***
*****
***                                     ***
***      ***** ADDITIONAL INFORMATION *****                                     ***
***                                     ***
***      NUMBER OF ENERGY GROUPS          27      USE LATTICE GEOMETRY          YES ***
***                                     ***
***      NO. OF FISSION SPECTRUM SOURCE GROUP 1      GLOBAL ARRAY NUMBER          30 ***
***                                     ***
***      NO. OF SCATTERING ANGLES IN XSECS    2      NUMBER OF UNITS IN THE GLOBAL X DIR.    1 ***
***                                     ***
***      ENTRIES/NEUTRON IN THE NEUTRON BANK 22      NUMBER OF UNITS IN THE GLOBAL Y DIR.    1 ***
***                                     ***
***      ENTRIES/NEUTRON IN THE FISSION BANK 15      NUMBER OF UNITS IN THE GLOBAL Z DIR.    4 ***
***                                     ***
***      NUMBER OF MIXTURES USED              7      USE A GLOBAL REFLECTOR          YES ***
***                                     ***
***      NUMBER OF BIAS ID'S USED             1      USE NESTED HOLES              NO ***
***                                     ***
***      NUMBER OF DIFFERENTIAL ALBEDOS USED  0      NUMBER OF HOLES              12 ***
***                                     ***
***      TOTAL INPUT GEOMETRY REGIONS         44      MAXIMUM HOLE NESTING LEVEL      1 ***
***                                     ***
***      NUMBER OF GEOMETRY REGIONS USED      44      USE NESTED ARRAYS              YES ***
***                                     ***
***      LARGEST GEOMETRY UNIT NUMBER         40      NUMBER OF ARRAYS USED          3 ***
***                                     ***
***      LARGEST ARRAY NUMBER                 30      MAXIMUM ARRAY NESTING LEVEL    2 ***
***                                     ***
***      +X BOUNDARY CONDITION                REFLECT    -X BOUNDARY CONDITION          REFLECT ***
***                                     ***
***      +Y BOUNDARY CONDITION                REFLECT    -Y BOUNDARY CONDITION          REFLECT ***
***                                     ***
***      +Z BOUNDARY CONDITION                PER      -Z BOUNDARY CONDITION          PER ***
***                                     ***
*****

```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

GENERATION	GENERATION K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING....ONLY	696 INDEPENDENT	FISSION POINTS WERE GENERATED			
1	6.16925E-01	7.50000E-02	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING....ONLY	662 INDEPENDENT	FISSION POINTS WERE GENERATED			
2	5.93491E-01	1.06167E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING....ONLY	728 INDEPENDENT	FISSION POINTS WERE GENERATED			
3	6.84356E-01	1.38167E-01	6.84356E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	6.26718E-01	1.68333E-01	6.55537E-01	2.88194E-02	0.00000E+00	0.00000E+00
5	6.46315E-01	2.00500E-01	6.52463E-01	1.69205E-02	0.00000E+00	0.00000E+00
6	6.22181E-01	2.31500E-01	6.44893E-01	1.41585E-02	0.00000E+00	0.00000E+00
7	6.14499E-01	2.62667E-01	6.38814E-01	1.25391E-02	0.00000E+00	0.00000E+00
8	6.30625E-01	2.93833E-01	6.37449E-01	1.03287E-02	0.00000E+00	0.00000E+00
9	5.98297E-01	3.25000E-01	6.31856E-01	1.03675E-02	0.00000E+00	0.00000E+00
10	6.34209E-01	3.57833E-01	6.32150E-01	8.98335E-03	0.00000E+00	0.00000E+00
11	6.72266E-01	3.90000E-01	6.36607E-01	9.09035E-03	0.00000E+00	0.00000E+00
12	6.70923E-01	4.22833E-01	6.40039E-01	8.82513E-03	0.00000E+00	0.00000E+00
13	6.12652E-01	4.54000E-01	6.37549E-01	8.36187E-03	0.00000E+00	0.00000E+00
14	5.95713E-01	4.86000E-01	6.34063E-01	8.39179E-03	0.00000E+00	0.00000E+00
15	6.55639E-01	5.17167E-01	6.35722E-01	7.89573E-03	0.00000E+00	0.00000E+00
16	6.32060E-01	5.48333E-01	6.35461E-01	7.31470E-03	0.00000E+00	0.00000E+00
17	6.36867E-01	5.79333E-01	6.35555E-01	6.81026E-03	0.00000E+00	0.00000E+00
18	6.03067E-01	6.10500E-01	6.33524E-01	6.68618E-03	0.00000E+00	0.00000E+00
19	6.23159E-01	6.42500E-01	6.32914E-01	6.31009E-03	0.00000E+00	0.00000E+00
20	5.77865E-01	6.72833E-01	6.29856E-01	6.68927E-03	0.00000E+00	0.00000E+00
21	6.18462E-01	7.03833E-01	6.29256E-01	6.35577E-03	0.00000E+00	0.00000E+00
22	5.87318E-01	7.36000E-01	6.27160E-01	6.38383E-03	0.00000E+00	0.00000E+00
23	6.10810E-01	7.67000E-01	6.26381E-01	6.12194E-03	0.00000E+00	0.00000E+00
24	6.48811E-01	7.99167E-01	6.27401E-01	5.92542E-03	0.00000E+00	0.00000E+00
25	6.25223E-01	8.30167E-01	6.27306E-01	5.66272E-03	0.00000E+00	0.00000E+00
26	6.38680E-01	8.61333E-01	6.27780E-01	5.44232E-03	0.00000E+00	0.00000E+00
27	6.22584E-01	8.92500E-01	6.27572E-01	5.22422E-03	0.00000E+00	0.00000E+00
28	6.47331E-01	9.24500E-01	6.28332E-01	5.07648E-03	0.00000E+00	0.00000E+00
29	6.29750E-01	9.56500E-01	6.28384E-01	4.88512E-03	0.00000E+00	0.00000E+00
30	6.02498E-01	9.86833E-01	6.27460E-01	4.79735E-03	0.00000E+00	0.00000E+00
31	6.50960E-01	1.01883E+00	6.28270E-01	4.69936E-03	0.00000E+00	0.00000E+00
32	6.05531E-01	1.05083E+00	6.27512E-01	4.60285E-03	0.00000E+00	0.00000E+00
33	6.41554E-01	1.08200E+00	6.27965E-01	4.47489E-03	0.00000E+00	0.00000E+00
34	6.21800E-01	1.11317E+00	6.27773E-01	4.33707E-03	0.00000E+00	0.00000E+00
35	6.43167E-01	1.14600E+00	6.28239E-01	4.22940E-03	0.00000E+00	0.00000E+00
36	6.63242E-01	1.17717E+00	6.29269E-01	4.23030E-03	0.00000E+00	0.00000E+00
37	6.25348E-01	1.20833E+00	6.29157E-01	4.10918E-03	0.00000E+00	0.00000E+00
38	6.20709E-01	1.24033E+00	6.28922E-01	4.00029E-03	0.00000E+00	0.00000E+00
39	6.47035E-01	1.27150E+00	6.29411E-01	3.92135E-03	0.00000E+00	0.00000E+00
40	6.04600E-01	1.30350E+00	6.28759E-01	3.87221E-03	0.00000E+00	0.00000E+00
41	6.40663E-01	1.33550E+00	6.29064E-01	3.78395E-03	0.00000E+00	0.00000E+00
42	6.42787E-01	1.36583E+00	6.29407E-01	3.70406E-03	0.00000E+00	0.00000E+00
43	6.11997E-01	1.39683E+00	6.28982E-01	3.63746E-03	0.00000E+00	0.00000E+00
44	6.26649E-01	1.42800E+00	6.28927E-01	3.55023E-03	0.00000E+00	0.00000E+00
.						
.						
.						
.						
974	6.37237E-01	3.07747E+01	6.28018E-01	6.60183E-04	0.00000E+00	0.00000E+00
975	6.36545E-01	3.08077E+01	6.28027E-01	6.59562E-04	0.00000E+00	0.00000E+00
976	6.11214E-01	3.08407E+01	6.28009E-01	6.59111E-04	0.00000E+00	0.00000E+00
977	6.09838E-01	3.08753E+01	6.27991E-01	6.58698E-04	0.00000E+00	0.00000E+00
978	6.06053E-01	3.09093E+01	6.27968E-01	6.58407E-04	0.00000E+00	0.00000E+00
979	6.15457E-01	3.09422E+01	6.27956E-01	6.57857E-04	0.00000E+00	0.00000E+00
980	6.20476E-01	3.09762E+01	6.27948E-01	6.57229E-04	0.00000E+00	0.00000E+00
981	6.27725E-01	3.10100E+01	6.27948E-01	6.56557E-04	0.00000E+00	0.00000E+00
982	6.23040E-01	3.10430E+01	6.27943E-01	6.55906E-04	0.00000E+00	0.00000E+00
983	6.02461E-01	3.10758E+01	6.27917E-01	6.55752E-04	0.00000E+00	0.00000E+00
984	6.34254E-01	3.11088E+01	6.27923E-01	6.55115E-04	0.00000E+00	0.00000E+00
985	6.56145E-01	3.11427E+01	6.27952E-01	6.55078E-04	0.00000E+00	0.00000E+00
986	6.42057E-01	3.11765E+01	6.27966E-01	6.54569E-04	0.00000E+00	0.00000E+00
987	6.26932E-01	3.12105E+01	6.27965E-01	6.53905E-04	0.00000E+00	0.00000E+00
988	6.04356E-01	3.12433E+01	6.27941E-01	6.53680E-04	0.00000E+00	0.00000E+00
989	6.26007E-01	3.12773E+01	6.27939E-01	6.53020E-04	0.00000E+00	0.00000E+00
990	6.38937E-01	3.13102E+01	6.27950E-01	6.52454E-04	0.00000E+00	0.00000E+00
991	6.52287E-01	3.13442E+01	6.27975E-01	6.52258E-04	0.00000E+00	0.00000E+00

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

992	5.99947E-01	3.13762E+01	6.27947E-01	6.52214E-04	0.00000E+00	0.00000E+00
993	6.57162E-01	3.14110E+01	6.27976E-01	6.52222E-04	0.00000E+00	0.00000E+00
994	6.25303E-01	3.14448E+01	6.27973E-01	6.51570E-04	0.00000E+00	0.00000E+00
995	6.57301E-01	3.14787E+01	6.28003E-01	6.51583E-04	0.00000E+00	0.00000E+00
996	6.58375E-01	3.15125E+01	6.28034E-01	6.51644E-04	0.00000E+00	0.00000E+00
997	6.38912E-01	3.15473E+01	6.28044E-01	6.51080E-04	0.00000E+00	0.00000E+00
998	6.69186E-01	3.15822E+01	6.28086E-01	6.51737E-04	0.00000E+00	0.00000E+00
999	5.99814E-01	3.16142E+01	6.28057E-01	6.51700E-04	0.00000E+00	0.00000E+00
1000	6.05547E-01	3.16472E+01	6.28035E-01	6.51437E-04	0.00000E+00	0.00000E+00
1001	6.35923E-01	3.16818E+01	6.28043E-01	6.50833E-04	0.00000E+00	0.00000E+00
1002	6.20651E-01	3.17148E+01	6.28035E-01	6.50224E-04	0.00000E+00	0.00000E+00
1003	6.54362E-01	3.17497E+01	6.28062E-01	6.50106E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123 EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.
FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

LIFETIME = 5.84212E-05 + OR - 6.19514E-08 GENERATION TIME = 5.82236E-05 + OR - 7.57390E-08
NU BAR = 2.43220E+00 + OR - 5.41433E-05 AVERAGE FISSION GROUP = 2.34261E+01 + OR - 3.14843E-03
ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 8.29065E-02 + OR - 2.34787E-04

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.62801	+ OR - 0.00065	0.62736 TO 0.62865	0.62671 TO 0.62930	0.62606 TO 0.62995	1000000
4	0.62801	+ OR - 0.00065	0.62736 TO 0.62866	0.62671 TO 0.62930	0.62606 TO 0.62995	999000
5	0.62799	+ OR - 0.00065	0.62734 TO 0.62864	0.62669 TO 0.62929	0.62604 TO 0.62994	998000
6	0.62799	+ OR - 0.00065	0.62734 TO 0.62864	0.62669 TO 0.62929	0.62604 TO 0.62994	997000
7	0.62801	+ OR - 0.00065	0.62736 TO 0.62866	0.62671 TO 0.62931	0.62606 TO 0.62996	996000
8	0.62801	+ OR - 0.00065	0.62735 TO 0.62866	0.62670 TO 0.62931	0.62605 TO 0.62996	995000
9	0.62803	+ OR - 0.00065	0.62738 TO 0.62869	0.62673 TO 0.62934	0.62608 TO 0.62999	994000
10	0.62803	+ OR - 0.00065	0.62738 TO 0.62868	0.62673 TO 0.62933	0.62607 TO 0.62998	993000
11	0.62798	+ OR - 0.00065	0.62733 TO 0.62863	0.62668 TO 0.62929	0.62603 TO 0.62994	992000
12	0.62794	+ OR - 0.00065	0.62729 TO 0.62859	0.62664 TO 0.62924	0.62599 TO 0.62989	991000
17	0.62795	+ OR - 0.00065	0.62730 TO 0.62860	0.62664 TO 0.62925	0.62599 TO 0.62990	986000
22	0.62808	+ OR - 0.00065	0.62743 TO 0.62873	0.62678 TO 0.62938	0.62613 TO 0.63003	981000
27	0.62807	+ OR - 0.00065	0.62742 TO 0.62873	0.62677 TO 0.62938	0.62611 TO 0.63004	976000
32	0.62808	+ OR - 0.00066	0.62742 TO 0.62873	0.62677 TO 0.62939	0.62611 TO 0.63004	971000
37	0.62802	+ OR - 0.00066	0.62736 TO 0.62868	0.62671 TO 0.62934	0.62605 TO 0.62999	966000
42	0.62801	+ OR - 0.00066	0.62735 TO 0.62867	0.62669 TO 0.62933	0.62603 TO 0.62998	961000
.						
.						
.						
.						
907	0.62920	+ OR - 0.00210	0.62709 TO 0.63130	0.62499 TO 0.63341	0.62288 TO 0.63551	96000
912	0.62889	+ OR - 0.00220	0.62669 TO 0.63109	0.62449 TO 0.63330	0.62229 TO 0.63550	91000
917	0.62841	+ OR - 0.00224	0.62617 TO 0.63065	0.62392 TO 0.63289	0.62168 TO 0.63513	86000
922	0.62901	+ OR - 0.00233	0.62668 TO 0.63135	0.62435 TO 0.63368	0.62202 TO 0.63601	81000
927	0.62951	+ OR - 0.00245	0.62706 TO 0.63196	0.62461 TO 0.63441	0.62216 TO 0.63686	76000
932	0.62985	+ OR - 0.00260	0.62725 TO 0.63245	0.62466 TO 0.63505	0.62206 TO 0.63765	71000
937	0.62965	+ OR - 0.00279	0.62687 TO 0.63244	0.62408 TO 0.63523	0.62129 TO 0.63801	66000

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

942	0.63042	+ OR - 0.00294	0.62748 TO 0.63336	0.62454 TO 0.63630	0.62160 TO 0.63924	61000
947	0.62861	+ OR - 0.00291	0.62570 TO 0.63152	0.62280 TO 0.63443	0.61989 TO 0.63733	56000
952	0.62923	+ OR - 0.00301	0.62623 TO 0.63224	0.62322 TO 0.63524	0.62022 TO 0.63825	51000
957	0.62924	+ OR - 0.00324	0.62600 TO 0.63247	0.62276 TO 0.63571	0.61953 TO 0.63895	46000
962	0.62837	+ OR - 0.00333	0.62504 TO 0.63170	0.62171 TO 0.63503	0.61837 TO 0.63837	41000
967	0.62847	+ OR - 0.00356	0.62491 TO 0.63203	0.62135 TO 0.63559	0.61779 TO 0.63914	36000
972	0.62969	+ OR - 0.00355	0.62614 TO 0.63323	0.62259 TO 0.63678	0.61904 TO 0.64033	31000
977	0.63072	+ OR - 0.00408	0.62664 TO 0.63480	0.62255 TO 0.63888	0.61847 TO 0.64297	26000
982	0.63362	+ OR - 0.00479	0.62882 TO 0.63841	0.62403 TO 0.64320	0.61924 TO 0.64799	21000
987	0.63400	+ OR - 0.00579	0.62821 TO 0.63979	0.62242 TO 0.64559	0.61663 TO 0.65138	16000
992	0.63841	+ OR - 0.00699	0.63142 TO 0.64540	0.62444 TO 0.65239	0.61745 TO 0.65938	11000
997	0.63091	+ OR - 0.01119	0.61972 TO 0.64210	0.60854 TO 0.65329	0.59735 TO 0.66448	6000

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

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                                FREQUENCY FOR GENERATIONS    4 TO 1003
0.5452 TO 0.5481.      *
0.5481 TO 0.5509
0.5509 TO 0.5537
0.5537 TO 0.5566
0.5566 TO 0.5594
0.5594 TO 0.5623
0.5623 TO 0.5651
0.5651 TO 0.5679      *
0.5679 TO 0.5708
0.5708 TO 0.5736      *
0.5736 TO 0.5765      *
0.5765 TO 0.5793      ****
0.5793 TO 0.5821      *****
0.5821 TO 0.5850      *****
0.5850 TO 0.5878      ****
0.5878 TO 0.5907      *****
0.5907 TO 0.5935      *****
0.5935 TO 0.5963      *****
0.5963 TO 0.5992      *****
0.5992 TO 0.6020      *****
0.6020 TO 0.6048      *****
0.6048 TO 0.6077      *****
0.6077 TO 0.6105      *****
0.6105 TO 0.6134      *****
0.6134 TO 0.6162      *****
0.6162 TO 0.6190      *****
0.6190 TO 0.6219      *****
0.6219 TO 0.6247      *****
0.6247 TO 0.6276      *****
0.6276 TO 0.6304      *****
0.6304 TO 0.6332      *****
0.6332 TO 0.6361      *****
0.6361 TO 0.6389      *****
0.6389 TO 0.6418      *****
0.6418 TO 0.6446      *****
0.6446 TO 0.6474      *****
0.6474 TO 0.6503      *****
0.6503 TO 0.6531      *****
0.6531 TO 0.6560      *****
0.6560 TO 0.6588      *****
0.6588 TO 0.6616      *****
0.6616 TO 0.6645      *****
0.6645 TO 0.6673      *****
0.6673 TO 0.6702      ****
0.6702 TO 0.6730      *****
0.6730 TO 0.6758      *****
0.6758 TO 0.6787      **
0.6787 TO 0.6815      **
0.6815 TO 0.6844      ****
0.6844 TO 0.6872
0.6872 TO 0.6900
0.6900 TO 0.6929
0.6929 TO 0.6957
0.6957 TO 0.6986
0.6986 TO 0.7014      *
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Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

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                                FREQUENCY FOR GENERATIONS 254 TO 1003
0.5452 TO 0.5481      *
0.5481 TO 0.5509
0.5509 TO 0.5537
0.5537 TO 0.5566
0.5566 TO 0.5594
0.5594 TO 0.5623
0.5623 TO 0.5651
0.5651 TO 0.5679      *
0.5679 TO 0.5708
0.5708 TO 0.5736      *
0.5736 TO 0.5765      *
0.5765 TO 0.5793      ***
0.5793 TO 0.5821      *****
0.5821 TO 0.5850      ****
0.5850 TO 0.5878      **
0.5878 TO 0.5907      *****
0.5907 TO 0.5935      *****
0.5935 TO 0.5963      *****
0.5963 TO 0.5992      *****
0.5992 TO 0.6020      *****
0.6020 TO 0.6048      *****
0.6048 TO 0.6077      *****
0.6077 TO 0.6105      *****
0.6105 TO 0.6134      *****
0.6134 TO 0.6162      *****
0.6162 TO 0.6190      *****
0.6190 TO 0.6219      *****
0.6219 TO 0.6247      *****
0.6247 TO 0.6276      *****
0.6276 TO 0.6304      *****
0.6304 TO 0.6332      *****
0.6332 TO 0.6361      *****
0.6361 TO 0.6389      *****
0.6389 TO 0.6418      *****
0.6418 TO 0.6446      *****
0.6446 TO 0.6474      *****
0.6474 TO 0.6503      *****
0.6503 TO 0.6531      *****
0.6531 TO 0.6560      *****
0.6560 TO 0.6588      *****
0.6588 TO 0.6616      *****
0.6616 TO 0.6645      *****
0.6645 TO 0.6673      *****
0.6673 TO 0.6702      ***
0.6702 TO 0.6730      **
0.6730 TO 0.6758      ****
0.6758 TO 0.6787      **
0.6787 TO 0.6815      **
0.6815 TO 0.6844      ***
0.6844 TO 0.6872
0.6872 TO 0.6900
0.6900 TO 0.6929
0.6929 TO 0.6957
0.6957 TO 0.6986
0.6986 TO 0.7014      *
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

FREQUENCY FOR GENERATIONS 504 TO 1003

```
0.5452 TO 0.5481
0.5481 TO 0.5509
0.5509 TO 0.5537
0.5537 TO 0.5566
0.5566 TO 0.5594
0.5594 TO 0.5623
0.5623 TO 0.5651
0.5651 TO 0.5679 *
0.5679 TO 0.5708
0.5708 TO 0.5736 *
0.5736 TO 0.5765 *
0.5765 TO 0.5793
0.5793 TO 0.5821 ****
0.5821 TO 0.5850 ***
0.5850 TO 0.5878 **
0.5878 TO 0.5907 *****
0.5907 TO 0.5935 *****
0.5935 TO 0.5963 *****
0.5963 TO 0.5992 *****
0.5992 TO 0.6020 *****
0.6020 TO 0.6048 *****
0.6048 TO 0.6077 *****
0.6077 TO 0.6105 *****
0.6105 TO 0.6134 *****
0.6134 TO 0.6162 *****
0.6162 TO 0.6190 *****
0.6190 TO 0.6219 *****
0.6219 TO 0.6247 *****
0.6247 TO 0.6276 *****
0.6276 TO 0.6304 *****
0.6304 TO 0.6332 *****
0.6332 TO 0.6361 *****
0.6361 TO 0.6389 *****
0.6389 TO 0.6418 *****
0.6418 TO 0.6446 *****
0.6446 TO 0.6474 *****
0.6474 TO 0.6503 *****
0.6503 TO 0.6531 *****
0.6531 TO 0.6560 *****
0.6560 TO 0.6588 *****
0.6588 TO 0.6616 *****
0.6616 TO 0.6645 *****
0.6645 TO 0.6673 *****
0.6673 TO 0.6702 **
0.6702 TO 0.6730 *
0.6730 TO 0.6758 ***
0.6758 TO 0.6787 **
0.6787 TO 0.6815 **
0.6815 TO 0.6844 *
0.6844 TO 0.6872
0.6872 TO 0.6900
0.6900 TO 0.6929
0.6929 TO 0.6957
0.6957 TO 0.6986
0.6986 TO 0.7014 *
```

Figure 6.7-7 CSAS25 Input / Output for Reconfigured Fuel Assembly Analysis (continued)

FAILED FUEL ASSEMBLY ANALYSIS - UNC TYPE FUEL - ZR/SS CLADS HOMOGENIZED

FREQUENCY FOR GENERATIONS 754 TO 1003

```
0.5452 TO 0.5481
0.5481 TO 0.5509
0.5509 TO 0.5537
0.5537 TO 0.5566
0.5566 TO 0.5594
0.5594 TO 0.5623
0.5623 TO 0.5651
0.5651 TO 0.5679
0.5679 TO 0.5708
0.5708 TO 0.5736
0.5736 TO 0.5765
0.5765 TO 0.5793
0.5793 TO 0.5821 **
0.5821 TO 0.5850 **
0.5850 TO 0.5878 *
0.5878 TO 0.5907 ***
0.5907 TO 0.5935 *****
0.5935 TO 0.5963 ***
0.5963 TO 0.5992 *****
0.5992 TO 0.6020 *****
0.6020 TO 0.6048 *****
0.6048 TO 0.6077 *****
0.6077 TO 0.6105 ****
0.6105 TO 0.6134 *****
0.6134 TO 0.6162 *****
0.6162 TO 0.6190 *****
0.6190 TO 0.6219 *****
0.6219 TO 0.6247 *****
0.6247 TO 0.6276 *****
0.6276 TO 0.6304 *****
0.6304 TO 0.6332 *****
0.6332 TO 0.6361 *****
0.6361 TO 0.6389 *****
0.6389 TO 0.6418 *****
0.6418 TO 0.6446 *****
0.6446 TO 0.6474 *****
0.6474 TO 0.6503 *****
0.6503 TO 0.6531 *****
0.6531 TO 0.6560 *****
0.6560 TO 0.6588 *****
0.6588 TO 0.6616 **
0.6616 TO 0.6645 *
0.6645 TO 0.6673 *****
0.6673 TO 0.6702 *
0.6702 TO 0.6730 *
0.6730 TO 0.6758 *
0.6758 TO 0.6787 *
0.6787 TO 0.6815 **
0.6815 TO 0.6844 *
0.6844 TO 0.6872
0.6872 TO 0.6900
0.6900 TO 0.6929
0.6929 TO 0.6957
0.6957 TO 0.6986
0.6986 TO 0.7014
```

Figure 6.7-8 CSAS25 Input/Output for Enlarged Fuel Tube Model

```
PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)

THIS IS A MODEL OF THE YNPS NAC-MPC BASKET
LOADED WITH 36 UNITED NUCLEAR TYPE A ASSEMBLIES
WITH MODIFIED TUBE ASSEMBLY 98
PRODUCED FOR THE YANKEE ROWE
STC LICENSE AMENDMENT

INTERIOR MODERATOR VOLUME FRACTION = 1.0
EXTERIOR MODERATOR VOLUME FRACTION = 1.0
CASK TO CASK PITCH = 300 cm
FLOODED PELLETT CLAD GAP
NEUTRON SHIELD REMOVED

27GROUPNDF4 LATTICECELL
UO2      1      0.95    293.0  92235 4.0 92238 96.0 END
ZIRCALLOY 2      1.0    293.0                      END
H2O      3      1.0    293.0                      END
AL       4      1.0    293.0                      END
SS304    5      1.0    293.0                      END
B-10     6  DEN=2.6226 0.0450 293.0                      END
B-11     6  DEN=2.6226 0.2736 293.0                      END
C        6  DEN=2.6226 0.0927 293.0                      END
AL       6  DEN=2.6226 0.5737 293.0                      END
PB       7      1.0    293.0                      END
H        8  DEN=1.6291 0.060  293.0                      END
O        8  DEN=1.6291 0.425  293.0                      END
C        8  DEN=1.6291 0.277  293.0                      END
N        8  DEN=1.6291 0.020  293.0                      END
AL       8  DEN=1.6291 0.214  293.0                      END
B-10     8  DEN=1.6291 0.001  293.0                      END
B-11     8  DEN=1.6291 0.004  293.0                      END
H2O      9      1.0    293.0                      END
H2O     10      1.0    293.0                      END
END COMP
SQUAREPITCH 1.1887 0.7887 1 3 0.9271 2 0.8052 10 END
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)
READ PARAM RUN=YES PLT=NO GEN=1003 NPG=1000 TME=500 END PARAM
READ GEOM

WATER LEVEL UNIT CELLS

UNIT 1
COM='FUEL PIN CELL - BETWEEN DISKS'
CYLINDER 1 1 0.3943 2P2.1400
CYLINDER 10 1 0.4026 2P2.1400
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 2
COM='WATER CELL - BETWEEN DISKS'
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 3
COM='DISPLACEMENT CELL - BETWEEN DISKS'
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 4
COM='INSTRUMENT TUBE CELL - BETWEEN DISKS'
CYLINDER 3 1 0.4998 2P2.1400
CYLINDER 5 1 0.5442 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400

DISK LEVEL UNIT CELLS (BOTH SS AND AL)

UNIT 5
COM='FUEL PIN CELL - WITH SS DISK'
CYLINDER 1 1 0.3943 2P0.6604
CYLINDER 10 1 0.4026 2P0.6604
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 6
COM='WATER CELL - WITH SS DISK'
CUBOID 3 1 4P0.5944 2P0.6604
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
UNIT 7
COM='DISPLACEMENT CELL - WITH SS DISK'
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 8
COM='INSTRUMENT TUBE CELL - WITH SS DISK'
CYLINDER 3 1 0.4998 2P0.6604
CYLINDER 5 1 0.5442 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604

' WATER LEVEL BORAL SHEETS
'
UNIT 14
COM='X-X BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P9.144 2P0.0318 2P2.1400
CUBOID 4 1 2P9.144 2P0.0953 2P2.1400
UNIT 15
COM='Y-Y BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P0.0318 2P9.144 2P2.1400
CUBOID 4 1 2P0.0953 2P9.144 2P2.1400
'
' DISK LEVEL BORAL SHEETS (AL AND SS)
'
UNIT 16
COM='X-X BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P9.144 2P0.0318 2P0.6604
CUBOID 4 1 2P9.144 2P0.0953 2P0.6604
UNIT 17
COM='Y-Y BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P0.0318 2P9.144 2P0.6604
CUBOID 4 1 2P0.0953 2P9.144 2P0.6604
'
' WATER LEVEL WEB MATERIAL
'
UNIT 20
COM='WATER LEVEL WEB MATERIAL (SMALL) X-X'
CUBOID 3 1 2P10.4635 2P0.9716 2P2.1400
UNIT 21
COM='WATER LEVEL WEB MATERIAL (MEDIUM) X-X'
CUBOID 3 1 2P10.4635 2P1.0478 2P2.1400
UNIT 22
COM='WATER LEVEL WEB MATERIAL (LARGE) X-X'
CUBOID 3 1 2P10.4635 2P1.1208 2P2.1400
UNIT 23
COM='WATER LEVEL WEB MATERIAL (LONG) Y-Y'
CUBOID 3 1 2P1.1208 2P79.5249 2P2.1400
'
' SUPPORT DISK WEB MATERIAL
'
UNIT 30
COM='SUPPORT DISK WEB MATERIAL (SMALL) X-X'
CUBOID 5 1 2P10.4635 2P0.9716 2P0.6604
UNIT 31
COM='SUPPORT DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 5 1 2P10.4635 2P1.0478 2P0.6604
UNIT 32
COM='SUPPORT DISK WEB MATERIAL (LARGE) X-X'
CUBOID 5 1 2P10.4635 2P1.1208 2P0.6604
UNIT 33
COM='SUPPORT DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 5 1 2P1.1208 2P79.5249 2P0.6604
'
' HEAT TRANSFER DISK WEB MATERIAL
'
UNIT 40
COM='HEAT TRANSFER DISK WEB MATERIAL (SMALL) X-X'
CUBOID 4 1 2P10.4635 2P0.9716 2P0.6604
UNIT 41
COM='HEAT TRANSFER DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 4 1 2P10.4635 2P1.0478 2P0.6604
UNIT 42
COM='HEAT TRANSFER DISK WEB MATERIAL (LARGE) X-X'
CUBOID 4 1 2P10.4635 2P1.1208 2P0.6604
UNIT 43
COM='HEAT TRANSFER DISK WEB MATERIAL (LONG) Y-Y'
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

CUBOID 4 1 2P1.1208 2P79.5249 2P0.6604
'
WATER LEVEL ASSEMBLY ARRAYS
'
UNIT 50
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL'
ARRAY 1 -9.5104 -9.5104 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400
UNIT 51
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.0 -0.1584 0.0
UNIT 52
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.0 0.1584 0.0
UNIT 53
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 0.0 0.0
UNIT 54
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 0.0 0.0
UNIT 55
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 0.1584 0.0
UNIT 56
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 0.1584 0.0
UNIT 57
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 -0.1584 0.0
UNIT 58
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 -0.1584 -0.1584 0.0
UNIT 59
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P2.1400
'
SUPPORT DISK LEVEL ASSEMBLY ARRAYS
'
UNIT 60
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
UNIT 61
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.0 -0.1584 0.0
UNIT 62
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.0 0.1584 0.0
UNIT 63
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 -0.1584 0.0 0.0

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

UNIT 64
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 0.0 0.0
UNIT 65
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 0.1584 0.0
UNIT 66
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 -0.1584 0.1584 0.0
UNIT 67
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 0.1584 -0.1584 0.0
UNIT 68
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 60 -0.1584 -0.1584 0.0
UNIT 69
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636                2P0.6604

HEAT TRANSFER DISK LEVEL ASSEMBLY ARRAYS
UNIT 70
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P9.9441                2P0.6604
CUBOID 5 1 4P10.0661                2P0.6604
CUBOID 3 1 4P10.25681               2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051               2P0.6604
UNIT 71
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.0 -0.1584 0.0
UNIT 72
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.0 0.1584 0.0
UNIT 73
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 -0.1584 0.0 0.0
UNIT 74
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.1584 0.0 0.0
UNIT 75
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.1584 0.1584 0.0
UNIT 76
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 -0.1584 0.1584 0.0
UNIT 77
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 0.1584 -0.1584 0.0
UNIT 78
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635                2P0.6604
HOLE 70 -0.1584 -0.1584 0.0
UNIT 79
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636                2P0.6604

WATER LEVEL BASKET ARRAYS
UNIT 80
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 20 -10.4636 -33.6323 -2.1400

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
UNIT 81
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 21 -10.4636 -33.6323 -2.1400
UNIT 82
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 22 -10.4636 -56.6549 -2.1400
UNIT 83
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 23 -10.4636 -56.6549 -2.1400
UNIT 84
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY-X)'
ARRAY 24 -10.4636 -79.5251 -2.1400
UNIT 85
COM='13X1 WATER LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 25 -10.4636 -79.5251 -2.1400
UNIT 86
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 26 -10.4636 -79.5251 -2.1400
'
SUPPORT DISK LEVEL BASKET ARRAYS
'
UNIT 90
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 30 -10.4636 -33.6323 -0.6604
UNIT 91
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 31 -10.4636 -33.6323 -0.6604
UNIT 92
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 32 -10.4636 -56.6549 -0.6604
UNIT 93
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 33 -10.4636 -56.6549 -0.6604
UNIT 94
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 34 -10.4636 -79.5251 -0.6604
UNIT 95
COM='13X1 SUPPORT DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 35 -10.4636 -79.5251 -0.6604
UNIT 96
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 36 -10.4636 -79.5251 -0.6604
'
HEAT TRANSFER DISK LEVEL BASKET ARRAYS
'
UNIT 100
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 40 -10.4636 -33.6323 -0.6604
UNIT 101
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 41 -10.4636 -33.6323 -0.6604
UNIT 102
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 42 -10.4636 -56.6549 -0.6604
UNIT 103
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 43 -10.4636 -56.6549 -0.6604
UNIT 104
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 44 -10.4636 -79.5251 -0.6604
UNIT 105
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 45 -10.4636 -79.5251 -0.6604
UNIT 106
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 46 -10.4636 -79.5251 -0.6604
'
BASKET ARRAY IN TRANSPORT CASK OVERPACK (LEVEL CONSTRUCTION)
UNIT 110
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - WATER LEVEL'
ARRAY 50 -33.6323 -79.5251 -2.1400
CYLINDER 3 1 88.1253 2P2.1400
HOLE 80 -69.0614 0.0 0.0
HOLE 82 -46.1912 0.0 0.0
HOLE 81 69.0614 0.0 0.0
HOLE 83 46.1912 0.0 0.0
```


Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
CYLINDER 5 1 89.7128      2P2.1400
CYLINDER 3 1 90.170       2P2.1400
CYLINDER 5 1 93.98        2P2.1400
CYLINDER 7 1 103.4288     2P2.1400
CYLINDER 5 1 110.109      2P2.1400
CYLINDER 9 1 124.714      2P2.1400
CYLINDER 9 1 125.349      2P2.1400
CUBOID 9 1 4P150          2P2.1400
UNIT 111
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - SUPPORT DISK LEVEL'
ARRAY 51 -33.6323 -79.5251 -0.6604
CYLINDER 5 1 87.6046      2P0.6604
HOLE 90 -69.0614 0.0 0.0
HOLE 92 -46.1912 0.0 0.0
HOLE 91 69.0614 0.0 0.0
HOLE 93 46.1912 0.0 0.0
CYLINDER 3 1 88.1253      2P0.6604
CYLINDER 5 1 89.7128      2P0.6604
CYLINDER 3 1 90.170       2P0.6604
CYLINDER 5 1 93.98        2P0.6604
CYLINDER 7 1 103.4288     2P0.6604
CYLINDER 5 1 110.109      2P0.6604
CYLINDER 9 1 124.714      2P0.6604
CYLINDER 9 1 125.349      2P0.6604
CUBOID 9 1 4P150          2P0.6604
UNIT 112
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - HEAT TRANSFER DISK LEVEL'
ARRAY 52 -33.6323 -79.5251 -0.6604
CYLINDER 4 1 87.2490      2P0.6604
HOLE 100 -69.0614 0.0 0.0
HOLE 102 -46.1912 0.0 0.0
HOLE 101 69.0614 0.0 0.0
HOLE 103 46.1912 0.0 0.0
CYLINDER 3 1 88.1253      2P0.6604
CYLINDER 5 1 89.7128      2P0.6604
CYLINDER 3 1 90.170       2P0.6604
CYLINDER 5 1 93.98        2P0.6604
CYLINDER 7 1 103.4288     2P0.6604
CYLINDER 5 1 110.109      2P0.6604
CYLINDER 9 1 124.714      2P0.6604
CYLINDER 9 1 125.349      2P0.6604
CUBOID 9 1 4P150          2P0.6604
UNIT 130
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- WATER LEVEL'
ARRAY 1 -9.5104 -9.5104 -2.1400
CUBOID 3 1 4P10.1473      2P2.1400
CUBOID 5 1 4P10.26922     2P2.1400
CUBOID 3 1 4P10.3051      2P2.1400
UNIT 131
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL +X +Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 130 0.1584 0.1584 0.0
UNIT 132
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL -X +Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 130 -0.1584 0.1584 0.0
UNIT 133
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL +X -Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 130 0.1584 -0.1584 0.0
UNIT 134
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL -X -Y'
CUBOID 3 1 4P10.4635      2P2.1400
HOLE 130 -0.1584 -0.1584 0.0
UNIT 140
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- SUPPORT DISK LEVEL'
ARRAY 2 -9.5104 -9.5104 -0.6604
CUBOID 3 1 4P10.1473      2P0.6604
CUBOID 5 1 4P10.26922     2P0.6604
CUBOID 3 1 4P10.3051      2P0.6604
UNIT 141
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL +X +Y'
CUBOID 3 1 4P10.4635      2P0.6604
HOLE 140 0.1584 0.1584 0.0
UNIT 142
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

[illegible]

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
WATER LEVEL ARRAYS
,
ARA=20  NUX=1  NUY=5  NUZ=1
FILL
55
22
54
22
57
END FILL
ARA=21  NUX=1  NUY=5  NUZ=1
FILL
56
22
53
22
58
END FILL
ARA=22  NUX=1  NUY=9  NUZ=1
FILL
131
21
55
22
54
22
57
21
133
END FILL
ARA=23  NUX=1  NUY=9  NUZ=1
FILL
132
21
56
22
53
22
58
21
134
END FILL
ARA=24  NUX=1  NUY=13  NUZ=1
FILL
55
20
55
21
55
22
54
22
57
21
57
20
57
END FILL
ARA=25  NUX=1  NUY=13  NUZ=1
FILL
52
20
52
21
52
22
59
22
51
21
51
20
51
END FILL
ARA=26  NUX=1  NUY=13  NUZ=1
FILL
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
56
20
56
21
56
22
53
22
58
21
58
20
58
END FILL
'
  SUPPOR DISK LEVEL ARRAYS
ARA=30  NUX=1  NUY=5  NUZ=1
FILL
65
32
64
32
67
END FILL
ARA=31  NUX=1  NUY=5  NUZ=1
FILL
66
32
63
32
68
END FILL
ARA=32  NUX=1  NUY=9  NUZ=1
FILL
141
31
65
32
64
32
67
31
143
END FILL
ARA=33  NUX=1  NUY=9  NUZ=1
FILL
142
31
66
32
63
32
68
31
144
END FILL
ARA=34  NUX=1  NUY=13  NUZ=1
FILL
65
30
65
31
65
32
64
32
67
31
67
30
67
END FILL
ARA=35  NUX=1  NUY=13  NUZ=1
FILL
62
30
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
62
31
62
32
69
32
61
31
61
30
61
END FILL
ARA=36 NUX=1 NUY=13 NUZ=1
FILL
66
30
66
31
66
32
63
32
68
31
68
30
68
END FILL
HEAT TRANSFER DISK LEVEL ARRAYS
ARA=40 NUX=1 NUY=5 NUZ=1
FILL
75
42
74
42
77
END FILL
ARA=41 NUX=1 NUY=5 NUZ=1
FILL
76
42
73
42
78
END FILL
ARA=42 NUX=1 NUY=9 NUZ=1
FILL
151
41
75
42
74
42
77
41
153
END FILL
ARA=43 NUX=1 NUY=9 NUZ=1
FILL
152
41
76
42
73
42
78
41
154
END FILL
ARA=44 NUX=1 NUY=13 NUZ=1
FILL
75
40
75
41
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

75
42
74
42
77
41
77
40
77
END FILL
ARA=45 NUX=1 NUY=13 NUZ=1
FILL
72
40
72
41
72
42
79
42
71
41
71
40
71
END FILL
ARA=46 NUX=1 NUY=13 NUZ=1
FILL
76
40
76
41
76
42
73
42
78
41
78
40
78
END FILL
'
' MAJOR ARRAYS
'
ARA=50 NUX=5 NUY=1 NUZ=1
FILL
84 23 85 23 86
END FILL
ARA=51 NUX=5 NUY=1 NUZ=1
FILL
94 33 95 33 96
END FILL
ARA=52 NUX=5 NUY=1 NUZ=1
FILL
104 43 105 43 106
END FILL
'
' GLOBAL ARRAY
'
ARA=60 NUX=1 NUY=1 NUZ=4
FILL
112
110
111
110
END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIR END BOUNDS
READ PLOT
SCR=YES PIC=MAT LPI=10
UAX=1.0 VDN=-1.0 NAX=1500
'
' WHOLE BASKET HORIZONTAL SLICES
TTL='BASKET X-Y CROSS SECTION AT Z= 0.635 HEAT TRANSFER DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 0.635

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

XLR= 130 YLR= -130 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 3.44 WATER LEVEL'
XUL= -130 YUL= 130 ZUL= 3.44
XLR= 130 YLR= -130 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 6.236 SS DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 6.236
XLR= 130 YLR= -130 ZLR= 6.236
UAX=1.0 VDN=-1.0 NAX=1500 END

HEAT TRANSFER DISK LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I HEAT TRANSFER DISK'
XUL= 12. YUL= 80 ZUL= 0.635
XLR= 80.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II HEAT TRANSFER DISK'
XUL= 12.0 YUL= -12.0 ZUL= 0.635
XLR= 80 YLR= -80 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III HEAT TRANSFER DISK'
XUL= -80.0 YUL= -12.0 ZUL= 0.635
XLR= -12.0 YLR= -80.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV HEAT TRANSFER DISK'
XUL= -80.0 YUL= 80.0 ZUL= 0.635
XLR= -12.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END

WATER LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I WATER LEVEL'
XUL= 12. YUL= 80 ZUL= 3.44
XLR= 80.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II WATER LEVEL'
XUL= 12.0 YUL= -12.0 ZUL= 3.44
XLR= 80 YLR= -80 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III WATER LEVEL'
XUL= -80.0 YUL= -12.0 ZUL= 3.44
XLR= -12.0 YLR= -80.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV WATER LEVEL'
XUL= -80.0 YUL= 80.0 ZUL= 3.44
XLR= -12.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END

VERTICAL SLICES

TTL='BASKET X-Z CROSS SECTION ALUMINUM LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 1.27
XLR= 90 YLR=0.4 ZLR= -.1
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION WATER LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 4.318
XLR= 90 YLR=0.4 ZLR= 1.27
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION SS LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 6.858
XLR= 90 YLR=0.4 ZLR= 5.588
UAX=1.0 WDN=-1.0 NAX=1500 END
TTL='BASKET X-Z CROSS SECTION ENTIRE MODEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 12
XLR= 90 YLR=0.4 ZLR= 0
UAX=1.0 WDN=-1.0 NAX=1500 END
END PLOT
END DATA

SECONDARY MODULE 000008 HAS BEEN CALLED.
MODULE 000008 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 0.77 (SECONDS).
SECONDARY MODULE 000002 HAS BEEN CALLED.
MODULE 000002 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 6.64 (SECONDS).
SECONDARY MODULE 000009 HAS BEEN CALLED.
MODULE 000009 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 1247.14 (SECONDS).
MODULE CSAS25 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 1257.46 (SECONDS).

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

CCCCCCCCCCCC SSSSSSSSSSS AAAAAAAAAA SSSSSSSSSSS 2222222222 555555555555
CCCCCCCCCCCC SSSSSSSSSSS AAAAAAAAAA SSSSSSSSSSS 22222222222 555555555555
CC          CC SS      SS AA      AA SS      SS 22      22 55
CC          SS      AA      AA SS      SS      22      55
CC          SS      AA      AA SS      SS      22      55
CC          SSSSSSSSSSS AAAAAAAAAA SSSSSSSSSSS      22 555555555555
CC          SSSSSSSSSSS AAAAAAAAAA SSSSSSSSSSS      22 555555555555
CC          SS      AA      AA SS      SS      22      55
CC          SS      AA      AA SS      SS      22      55
CC          CC SS      SS AA      AA SS      SS 22      55      55
CCCCCCCCCCCC SSSSSSSSSSS AA      AA SSSSSSSSSSS 22222222222 555555555555
CCCCCCCCCCCC SSSSSSSSSSS AA      AA SSSSSSSSSSS 22222222222 55555555555

```

```

SSSSSSSSSS CCCCCCCCCC AAAAAAAAAA LL EEEEEEEEEEE PPPPPPPPPPP CCCCCCCCCC
SSSSSSSSSS CCCCCCCCCC AAAAAAAAAA LL EEEEEEEEEEE PPPPPPPPPPP CCCCCCCCCC
SS          CC          CC AA      AA LL EE          PP          PP CC          CC
SS          CC          AA AA      AA LL EE          PP          PP CC          CC
SS          CC          AA AA      AA LL EE          PP          PP CC          CC
SSSSSSSSSS CC          AAAAAAAAAA LL EEEEEEEEEEE ----- PPPPPPPPPPP CC
SSSSSSSSSS CC          AAAAAAAAAA LL EEEEEEEEEEE ----- PPPPPPPPPPP CC
          SS          CC AA      AA LL EE          PP          CC
          SS          CC AA      AA LL EE          PP          CC
SS          CC          CC AA      AA LL EE          PP          CC
SSSSSSSSSS CCCCCCCCCC AA      AA LLLLLLLLLLLLL EEEEEEEEEEE PP          CCCCCCCCCC
SSSSSSSSSS CCCCCCCCCC AA      AA LLLLLLLLLLLLL EEEEEEEEEEE PP          CCCCCCCCCC

```

```

00000000 8888888888 // 11 55555555555 // 00000000 00000000
000000000 888888888888 // 111 55555555555 // 000000000 000000000
00          00 88      88 // 1111 55 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
00          00 8888888888 // 11 5555555555 // 00          00 00      00
00          00 8888888888 // 11 5555555555 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
00          00 88      88 // 11 55 // 00          00 00      00
000000000 888888888888 // 11111111 55555555555 // 000000000 000000000
00000000 88888888888 // 11111111 5555555555 // 00000000 00000000

```

```

00000000 8888888888 44 77777777777 44 44
000000000 888888888888 444 77777777777 444 444
00          00 88      88 ::: 4444 77 ::: 4444 4444
00          00 88      88 ::: 44 44 77 ::: 44 44 44 44
00          00 88      88 ::: 44 44 77 ::: 44 44 44 44
00          00 8888888888 44 44 77 44 44 44 44
00          00 8888888888 44 44 77 44 44 44 44
00          00 88      88 ::: 444444444444 77 ::: 444444444444 444444444444
00          00 88      88 ::: 444444444444 77 ::: 444444444444 444444444444
00          00 88      88 ::: 44 77 ::: 44 44 44
000000000 888888888888 44 77 44 44
00000000 88888888888 44 77 44 44

```


SSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE		PPPPPPPPPP	CCCCCCCCCC						
SSSSSSSSSSSS	CCCCCCCCCCCC	AAAAAAAAAA	LL	EEEEEEEEEEEE		PPPPPPPPPPPP	CCCCCCCCCCCC						
SS	SS	CC	CC	AA	AA	LL	PP	PP	CC	CC			
SS		CC		AA		AA	LL	EE		PP	PP	CC	
SS		CC		AA		AA	LL	EE		PP		PP	CC
SSSSSSSSSSS	CC			AAAAAAAAAAAA	LL			EEEEEEEE	-----	PPPPPPPPPP	CC		
SSSSSSSSSSSS	CC			AAAAAAAAAAAA	LL			EEEEEEEE	-----	PPPPPPPPPP	CC		
	SS	CC		AA		AA	LL	EE		PP		CC	
	SS		CC	AA		AA	LL	EE		PP		CC	
SS	SS	CC	CC	AA		AA	LL	EE		PP		CC	CC
SSSSSSSSSSSS	CCCCCCCCCCCC	AA		AA		LLLLLLLLLLLL		EEEEEEEEEEEE		PP		CCCCCCCCCCCC	
SSSSSSSSSSS	CCCCCCCCCC	AA		AA		LLLLLLLLLLLL		EEEEEEEEEEEE		PP		CCCCCCCCCC	

[illegible]

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
**** PROBLEM PARAMETERS ****

LIB 27GROUPNDF4  LIBRARY
MXX             10 MIXTURES
MSC             19 COMPOSITION SPECIFICATIONS
IZM             4 MATERIAL ZONES
GE LATTICECELL  GEOMETRY
MORE           0 0/1 DO NOT READ/READ OPTIONAL PARAMETER DATA
MSLN           0 FUEL SOLUTIONS

**** PROBLEM COMPOSITION DESCRIPTION ****

SC UO2          STANDARD COMPOSITION
MX             1 MIXTURE NO.
VF            0.9500 VOLUME FRACTION
ROTH          10.9600 THEORETICAL DENSITY
NEL           2 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              92000      1.00 ATOM/MOLECULE
                      92235      4.000 WT%
                      92238      96.000 WT%
              8016      2.00 ATOMS/MOLECULE
END

SC ZIRCALLOY    STANDARD COMPOSITION
MX             2 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          6.5600 THEORETICAL DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              40302      1.00 ATOM/MOLECULE
END

SC H2O          STANDARD COMPOSITION
MX             3 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          0.9982 THEORETICAL DENSITY
NEL           2 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              1001      2.00 ATOMS/MOLECULE
              8016      1.00 ATOM/MOLECULE
END

SC AL           STANDARD COMPOSITION
MX             4 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          2.7020 THEORETICAL DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              13027      1.00 ATOM/MOLECULE
END

SC SS304        STANDARD COMPOSITION
MX             5 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          7.9200 THEORETICAL DENSITY
NEL           4 NO. ELEMENTS
ICP           0 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              24304      19.000 WT%
              25055      2.000 WT%
              26304      69.500 WT%
              28304      9.500 WT%
END

SC B-10         STANDARD COMPOSITION
MX             6 MIXTURE NO.
VF            0.0450 VOLUME FRACTION
ROTH          2.6226 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
5010      1.00 ATOM/MOLECULE
.END

SC B-11      STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.2736 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
5011      1.00 ATOM/MOLECULE
.END

SC C          STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.0927 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
6012      1.00 ATOM/MOLECULE
.END

SC AL         STANDARD COMPOSITION
MX           6 MIXTURE NO.
VF           0.5737 VOLUME FRACTION
ROTH         2.6226 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
13027      1.00 ATOM/MOLECULE
.END

SC PB         STANDARD COMPOSITION
MX           7 MIXTURE NO.
VF           1.0000 VOLUME FRACTION
ROTH         11.3440 THEORETICAL DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
82000      1.00 ATOM/MOLECULE
.END

SC H          STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.0600 VOLUME FRACTION
ROTH         1.6291 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
1001      1.00 ATOM/MOLECULE
.END

SC O          STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.4250 VOLUME FRACTION
ROTH         1.6291 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
8016      1.00 ATOM/MOLECULE
.END

SC C          STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.2770 VOLUME FRACTION
ROTH         1.6291 SPECIFIED DENSITY
NEL          1 NO. ELEMENTS
ICP          1 0/1 MIXTURE/COMPOUND
TEMP         293.0 DEG KELVIN
6012      1.00 ATOM/MOLECULE
.END

SC N          STANDARD COMPOSITION
MX           8 MIXTURE NO.
VF           0.0200 VOLUME FRACTION
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

ROTH      1.6291 SPECIFIED DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           7014      1.00 ATOM/MOLECULE
END

SC AL      STANDARD COMPOSITION
MX          8 MIXTURE NO.
VF        0.2140 VOLUME FRACTION
ROTH      1.6291 SPECIFIED DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           13027     1.00 ATOM/MOLECULE
END

SC B-10    STANDARD COMPOSITION
MX          8 MIXTURE NO.
VF        0.0010 VOLUME FRACTION
ROTH      1.6291 SPECIFIED DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           5010     1.00 ATOM/MOLECULE
END

SC B-11    STANDARD COMPOSITION
MX          8 MIXTURE NO.
VF        0.0040 VOLUME FRACTION
ROTH      1.6291 SPECIFIED DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           5011     1.00 ATOM/MOLECULE
END

SC H2O     STANDARD COMPOSITION
MX          9 MIXTURE NO.
VF        1.0000 VOLUME FRACTION
ROTH      0.9982 THEORETICAL DENSITY
NEL        2 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           1001     2.00 ATOMS/MOLECULE
           8016     1.00 ATOM/MOLECULE
END

SC H2O     STANDARD COMPOSITION
MX          10 MIXTURE NO.
VF        1.0000 VOLUME FRACTION
ROTH      0.9982 THEORETICAL DENSITY
NEL        2 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
           1001     2.00 ATOMS/MOLECULE
           8016     1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****
CTP SQUAREPITCH CELL TYPE
PITCH      1.1887 CM CENTER TO CENTER SPACING
FUELOD     0.7887 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL      1 MIXTURE NO. OF FUEL
MMOD       3 MIXTURE NO. OF MODERATOR
CLADOD     0.9271 CM CLAD OUTER DIAMETER
MCLAD      2 MIXTURE NO. OF CLAD
GAPOD      0.8052 CM GAP OUTER DIAMETER
MGAP       10 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```
*****
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)***
***
*****
***
***          ***** DATA LIBRARY INFORMATION *****
***
***          UNIT          DATA SET NAME          VOLUME          UNIT FUNCTION
***          NUMBER          NAME          NAME          -----
***          -----
***          89          G:\scale43\DALALIB\FT89F001          STANDARD COMPOSITION LIBRARY
***          82          G:\scale43\DALALIB\FT82F001          CROSS SECTION LIBRARY
***          11          D:\zjr\Yankee\tpa300-F_r1\FT11F001          SHORT CROSS SECTION LIBRARY
***          90          D:\zjr\Yankee\tpa300-F_r1\FT90F001          INPUT DATA DIRECT ACCESS
***
*****
***
***          STANDARD COMPOSITION LIBRARY DATA
***          -----
***
***          UNIT NUMBER : 89
***
***          DATASET NAME : G:\scale43\DALALIB\FT89F001
***
***          LIBRARY TITLE: SCALE-4 STANDARD COMPOSITION LIBRARY
***          637 STANDARD COMPOSITIONS, 490 NUCLIDES
***          90 ELEMENTS WITH VARIABLE ISOTOPIC DISTRUBUTIONS.
***
***          CREATION DATE: 6/30/95
***
***
***          CROSS SECTION LIBRARY DATA
***          -----
***
***          UNIT NUMBER : 82
***
***          DATASET NAME : G:\scale43\DALALIB\FT82F001
***
***          LIBRARY TITLE: SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
***          BASED ON ENDF-B VERSION 4 DATA
***          COMPILED FOR NRC          1/27/89
***          LAST UPDATED
***          L.M.PETRIE - ORNL
***          08/12/94
***
*****
*****
*****          0 IO'S WERE USED BEFORE READING KENO V DATA          *****
*****          0 IO'S WERE USED READING THE KENO V PARAMETER DATA          *****
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

KK	KK	EEEEEEEEEEEE	NN	NN	0000000000	VV	VV
KK	KK	EEEEEEEEEEEE	NNN	NN	000000000000	VV	VV
KK	KK	EE	NNNN	NN	00	VV	VV
KK	KK	EE	NN NN	NN	00	VV	VV
KK	KK	EE	NN NN	NN	00	VV	VV
KKKKKKKK	EEEEEEEE	NN	NN	NN	00	VV	VV
KKKKKKKK	EEEEEEEE	NN	NN	NN	00	VV	VV
KK	KK	EE	NN	NN NN	00	VV	VV
KK	KK	EE	NN	NN NN	00	VV	VV
KK	KK	EE	NN	NNNN	00	VV	VV
KK	KK	EEEEEEEEEEEE	NN	NNN	000000000000	VV	VV
KK	KK	EEEEEEEEEEEE	NN	NN	0000000000	V	

SSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SSSSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SSSSSSSSSS	CC	AAAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SSSSSSSSSS	CC	AAAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	PP	CCCCCCCCCC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	PP	CCCCCCCCCC

0000000	8888888888	//	11	5555555555	//	0000000	0000000
000000000	888888888888	//	111	555555555555	//	000000000	000000000
00	00	88	88	55	//	00	00
00	00	88	88	55	//	00	00
00	00	88	88	55	//	00	00
00	00	8888888888	11	5555555555	//	00	00
00	00	8888888888	11	555555555555	//	00	00
00	00	88	88	55	//	00	00
00	00	88	88	55	//	00	00
00	00	88	88	55	//	00	00
00	00	88	88	55	//	00	00
000000000	888888888888	//	1111111	555555555555	//	000000000	000000000
0000000	88888888888	//	1111111	5555555555	//	0000000	0000000

0000000	88888888888		44	77777777777		55555555555	3333333333
000000000	888888888888		444	77777777777		555555555555	333333333333
00	00	88	88	77		55	33
00	00	88	88	77		55	33
00	00	88	88	77		55	33
00	00	8888888888	44	77		55555555555	333
00	00	8888888888	44	77		555555555555	333
00	00	88	88	77		55	33
00	00	88	88	77		55	33
00	00	88	88	77		55	33
000000000	888888888888		44	77		555555555555	333333333333
0000000	88888888888		44	77		55555555555	3333333333

SSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SSSSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SSSSSSSSSS	CC	AAAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SSSSSSSSSS	CC	AAAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SS	CC	AA	AA	EE	PP	CC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	PP	CCCCCCCCCC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	PP	CCCCCCCCCC

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

*****
*****      PROGRAM VERIFICATION INFORMATION      *****
*****      CODE SYSTEM:  SCALE-PC VERSION:  4.3      *****
*****
*****      PROGRAM:  000009      *****
*****      CREATION DATE:  03/08/96      *****
*****      VOLUME:  ENG      *****
*****      LIBRARY:  G:\SCALE43\WIN_NT\EXE      *****
*****      PRODUCTION CODE:  KENOVA      *****
*****      VERSION:  3.1      *****
*****      JOBNAME:  SCALE-PC      *****
*****      DATE OF EXECUTION:  08/15/00      *****
*****      TIME OF EXECUTION:  08:47:53      *****
*****

*****
*****      TRANSPORT CRITICALITY:  NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)*****
*****
*****      *****      NUMERIC PARAMETERS      *****
*****
*****      TME      MAXIMUM PROBLEM TIME (MIN)      500.00      *****
*****      TBA      TIME PER GENERATION (MIN)      0.50      *****
*****      GEN      NUMBER OF GENERATIONS      1003      *****
*****      NPG      NUMBER PER GENERATION      1000      *****
*****      NSK      NUMBER OF GENERATIONS TO BE SKIPPED      3      *****
*****      BEG      BEGINNING GENERATION NUMBER      1      *****
*****      RES      GENERATIONS BETWEEN CHECKPOINTS      0      *****
*****      X1D      NUMBER OF EXTRA 1-D CROSS SECTIONS      1      *****
*****      NBK      NEUTRON BANK SIZE      1025      *****
*****      XNB      EXTRA POSITIONS IN NEUTRON BANK      0      *****
*****      NFB      FISSION BANK SIZE      1000      *****
*****      XFB      EXTRA POSITIONS IN FISSION BANK      0      *****
*****      WTA      DEFAULT VALUE OF WEIGHT AVERAGE      0.5000      *****
*****      WTH      WEIGHT HIGH FOR SPLITTING      3.0000      *****
*****      WTL      WEIGHT LOW FOR RUSSIAN ROULETTE      0.3333      *****
*****      RND      STARTING RANDOM NUMBER      BB827100001      *****
*****      NB8      NUMBER OF D.A. BLOCKS ON UNIT  8      200      *****
*****      NL8      LENGTH OF D.A. BLOCKS ON UNIT  8      512      *****
*****      ADJ      MODE OF CALCULATION      FORWARD      *****
*****      INPUT DATA WRITTEN ON RESTART UNIT      NO      *****
*****      BINARY DATA INTERFACE      YES      *****
*****

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

*****
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
***
*****
***          LOGICAL PARAMETERS          ***
***
*** RUN  EXECUTE PROBLEM AFTER CHECKING DATA  YES      PLT  PLOT PICTURE MAP(S)          NO ***
*** FLX  COMPUTE FLUX                          NO      FDN  COMPUTE FISSION DENSITIES        NO ***
*** SMU  COMPUTE AVG UNIT SELF-MULTIPLICATION  NO      NUB  COMPUTE NU-BAR & AVG FISSION GROUP  YES ***
*** MKU  COMPUTE MATRIX K-EFF BY UNIT NUMBER   NO      MKP  COMPUTE MATRIX K-EFF BY UNIT LOCATION NO ***
*** CKU  COMPUTE COFACTOR K-EFF BY UNIT NUMBER NO      CKP  COMPUTE COFACTOR K-EFF BY UNIT LOCATION NO ***
*** FMU  PRINT FISSION PROD MATRIX BY UNIT NUMBER NO  FMP  PRINT FISSION PROD MATRIX BY UNIT LOCATION NO ***
*** MKH  COMPUTE MATRIX K-EFF BY HOLE NUMBER   NO      MKA  COMPUTE MATRIX K-EFF BY ARRAY NUMBER  NO ***
*** CKH  COMPUTE COFACTOR K-EFF BY HOLE NUMBER NO      CKA  COMPUTE COFACTOR K-EFF BY ARRAY NUMBER  NO ***
*** FMH  PRINT FISSION PROD MATRIX BY HOLE NUMBER NO  FMA  PRINT FISSION PROD MATRIX BY ARRAY NUMBER  NO ***
*** HHL  COLLECT MATRIX BY HIGHEST HOLE LEVEL  NO      HAL  COLLECT MATRIX BY HIGHEST ARRAY LEVEL  NO ***
*** AMX  PRINT ALL MIXED CROSS SECTIONS        NO      FAR  PRINT FIS. AND ABS. BY REGION        NO ***
*** XS1  PRINT 1-D MIXTURE X-SECTIONS          NO      GAS  PRINT FAR BY GROUP                NO ***
*** XS2  PRINT 2-D MIXTURE X-SECTIONS          NO      PAX  PRINT XSEC-ALBEDO CORRELATION TABLES NO ***
*** XAP  PRINT MIXTURE ANGLES & PROBABILITIES  NO      PWT  PRINT WEIGHT AVERAGE ARRAY        NO ***
*** PKI  PRINT FISSION SPECTRUM                NO      PGM  PRINT INPUT GEOMETRY              NO ***
*** P1D  PRINT EXTRA 1-D CROSS SECTIONS        NO      BUG  PRINT DEBUG INFORMATION          NO ***
***                                     TRK  PRINT TRACKING INFORMATION        NO ***
***
*****
***          PARAMETER INPUT COMPLETED
***          0 IO'S WERE USED READING THE PARAMETER DATA
***          ***** DATA READING COMPLETED *****
*****
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
***
*****
***          UNIT          DATA SET NAME          VOLUME          UNIT FUNCTION
***          NUMBER          DATA SET NAME          NAME          DATA SET NAME
***          -----          -----          -----          -----
***          XSC  14      D:\zjr\Yankee\tpa300-F_r1\FT14F001      MIXED CROSS SECTIONS
***          ALB  79      G:\scale43\DATALIB\FT79F001      INPUT ALBEDOS
***          WTS  80      G:\scale43\DATALIB\FT80F001      INPUT WEIGHTS
***          SKT  16      UNKNOWN      WRITE SCRATCH DATA
***          BIN  95      D:\zjr\Yankee\tpa300-F_r1\FT95F001      BINARY INPUT DATA
***          RST  95      D:\zjr\Yankee\tpa300-F_r1\FT95F001      READ RESTART DATA
***          LIB  4      D:\zjr\Yankee\tpa300-F_r1\FT04F001      INPUT AMPX WORKING LIBRARY
***          8      D:\zjr\Yankee\tpa300-F_r1\FT08F001      INPUT DATA DIRECT ACCESS
***          9      UNKNOWN      SUPER GROUPED DIRECT ACCESS
***          10     UNKNOWN      XSEC MIXING DIRECT ACCESS
***
*****
***          0 IO'S WERE USED PREPARING INPUT DATA
***          CROSS SECTIONS READ FROM THE AMPX WORKING LIBRARY ON UNIT 4

```


Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD = 3.0E-05

MIXTURE = 1		DENSITY(G/CC) = 10.412					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
1008016	4.64617E-02	1.18487E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
1092235	9.40641E-04	3.52606E-02	92235	235.0441	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED
08/12/94							
1092238	2.22902E-02	8.46253E-01	92238	238.0510	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED
08/12/94							
MIXTURE = 2		DENSITY(G/CC) = 6.5600					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
2040302	4.33078E-02	1.00000E+00	40000	91.2196	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED
08/12/94							
MIXTURE = 3		DENSITY(G/CC) = 0.99817					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
3001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
3008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							
MIXTURE = 4		DENSITY(G/CC) = 2.7020					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
4013027	6.03066E-02	1.00000E+00	13027	26.9818	AL-27 1193 218 GP 040375(5)		UPDATED
08/12/94							
MIXTURE = 5		DENSITY(G/CC) = 7.9200					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
5024304	1.74286E-02	1.90000E-01	24000	51.9957	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
5025055	1.73633E-03	1.99999E-02	25055	54.9379	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED
08/12/94							
5026304	5.93579E-02	6.95000E-01	26000	55.8447	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
5028304	7.72070E-03	9.50001E-02	28000	58.6872	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)		UPDATED
08/12/94							
MIXTURE = 6		DENSITY(G/CC) = 2.5833					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
6005010	7.09799E-03	4.56855E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K		UPDATED
08/12/94							
6005011	3.92499E-02	2.77771E-01	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160	UPDATED
08/12/94							
6006012	1.22006E-02	9.41116E-02	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94							
6013027	3.35812E-02	5.82432E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)		UPDATED
08/12/94							
MIXTURE = 7		DENSITY(G/CC) = 11.344					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K		UPDATED
08/12/94							
MIXTURE = 8		DENSITY(G/CC) = 1.6307					
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE		
8001001	5.84084E-02	5.99323E-02	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94							
8005010	9.79802E-05	9.99025E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K		UPDATED
08/12/94							
8005011	3.56450E-04	3.99615E-03	5011	11.0096	BORON-11	ENDF/B-IV MAT 1160	UPDATED
08/12/94							
8006012	2.26463E-02	2.76729E-01	6000	12.0001	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94							
8007014	1.40121E-03	1.99805E-02	7014	14.0033	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED
08/12/94							
8008016	2.60749E-02	4.24574E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED
08/12/94							

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

8013027	7.78110E-03	2.13789E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE =	9	DENSITY(G/CC) =	0.99817			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
9001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002
08/12/94						
9008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276
08/12/94						
MIXTURE =	10	DENSITY(G/CC) =	0.99817			
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
10001001	6.67692E-02	1.11927E-01	1001	1.0077	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002
08/12/94						
10008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16	ENDF/B-IV MAT 1276
08/12/94						
3001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED	08/12/94		
8001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED	08/12/94		
9001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED	08/12/94		
10001001	HYDROGEN	ENDF/B-IV MAT 1269/THRM1002	UPDATED	08/12/94		
6005010	B-10 1273 218NGP 042375 P-3 293K	UPDATED	08/12/94			
8005010	B-10 1273 218NGP 042375 P-3 293K	UPDATED	08/12/94			
6005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED	08/12/94		
8005011	BORON-11	ENDF/B-IV MAT 1160	UPDATED	08/12/94		
6006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED	08/12/94		
8006012	CARBON-12	ENDF/B-IV MAT 1274/THRM1065	UPDATED	08/12/94		
8007014	NITROGEN-14	ENDF/B-IV MAT 1275	UPDATED	08/12/94		
1008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED	08/12/94		
3008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED	08/12/94		
8008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED	08/12/94		
9008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED	08/12/94		
10008016	OXYGEN-16	ENDF/B-IV MAT 1276	UPDATED	08/12/94		
4013027	AL-27 1193 218 GP 040375(5)	UPDATED	08/12/94			
6013027	AL-27 1193 218 GP 040375(5)	UPDATED	08/12/94			
8013027	AL-27 1193 218 GP 040375(5)	UPDATED	08/12/94			
5024304	CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED	08/12/94			
5025055	MANGANESE-55	ENDF/B-IV MAT 1197	UPDATED	08/12/94		
5026304	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED	08/12/94			
5028304	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED	08/12/94			
2040302	ZIRCALLOY	ENDF/B-IV MAT 1284	UPDATED	08/12/94		
7082000	PB 1288 218NGP 042375 P-3 293K	UPDATED	08/12/94			
1092235	URANIUM-235	ENDF/B-IV MAT 1261	UPDATED	08/12/94		
1092238	URANIUM-238	ENDF/B-IV MAT 1262	UPDATED	08/12/94		
KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 3 WERE CORRECTED FOR BAD MOMENTS.						
KENO MESSAGE NUMBER K5-222 2 TRANSFERS FOR MIXTURE 8 WERE CORRECTED FOR BAD MOMENTS.						
KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 9 WERE CORRECTED FOR BAD MOMENTS.						
KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 10 WERE CORRECTED FOR BAD MOMENTS.						
..... 0 IO'S WERE USED MIXING CROSS-SECTIONS						
1-D CROSS SECTION ARRAY ID NUMBERS						
1 2002 1452 27 18 1018						
..... 0 IO'S WERE USED PREPARING THE CROSS SECTIONS						

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

```

*****
***
***      TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0) ***
***
*****
***
***      ***** ADDITIONAL INFORMATION *****
***
***      NUMBER OF ENERGY GROUPS          27      USE LATTICE GEOMETRY          YES ***
***
***      NO. OF FISSION SPECTRUM SOURCE GROUP 1      GLOBAL ARRAY NUMBER          60 ***
***
***      NO. OF SCATTERING ANGLES IN XSECS    2      NUMBER OF UNITS IN THE GLOBAL X DIR.  1 ***
***
***      ENTRIES/NEUTRON IN THE NEUTRON BANK  33      NUMBER OF UNITS IN THE GLOBAL Y DIR.  1 ***
***
***      ENTRIES/NEUTRON IN THE FISSION BANK  26      NUMBER OF UNITS IN THE GLOBAL Z DIR.  4 ***
***
***      NUMBER OF MIXTURES USED              9      USE A GLOBAL REFLECTOR          YES ***
***
***      NUMBER OF BIAS ID'S USED             1      USE NESTED HOLES                YES ***
***
***      NUMBER OF DIFFERENTIAL ALBEDOS USED  0      NUMBER OF HOLES                60 ***
***
***      TOTAL INPUT GEOMETRY REGIONS         160     MAXIMUM HOLE NESTING LEVEL       3 ***
***
***      NUMBER OF GEOMETRY REGIONS USED      160     USE NESTED ARRAYS                YES ***
***
***      LARGEST GEOMETRY UNIT NUMBER         154     NUMBER OF ARRAYS USED           27 ***
***
***      LARGEST ARRAY NUMBER                 60      MAXIMUM ARRAY NESTING LEVEL     4 ***
***
***      +X BOUNDARY CONDITION                MIR     -X BOUNDARY CONDITION          MIR ***
***
***      +Y BOUNDARY CONDITION                MIR     -Y BOUNDARY CONDITION          MIR ***
***
***      +Z BOUNDARY CONDITION                PER     -Z BOUNDARY CONDITION          PER ***
***
*****
***
***      TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)
***
***      TOTAL MIXTURE VOLUMES
***
***      MIXTURE      TOTAL VOLUME      MASS(G)
***      1            4.66804E+04 CM**3    4.86035E+05
***      2            1.63811E+04 CM**3    1.07460E+05
***      3            1.80087E+05 CM**3    1.79757E+05
***      4            1.35152E+04 CM**3    3.65179E+04
***      5            1.00696E+05 CM**3    7.97513E+05
***      6            1.66768E+03 CM**3    4.30809E+03
***      7            6.56407E+04 CM**3    7.44628E+05
***      9            5.81490E+05 CM**3    5.80427E+05
***      10           1.98592E+03 CM**3    1.98228E+03
***
*****
***
***      BIASING INFORMATION
***
***      A DEFAULT WEIGHT OF 0.500 WILL BE USED FOR ALL BIAS ID'S.
***
*****
***
***      0 IO'S WERE USED IN KENO-V BEFORE TRACKING
***
***      0.01383 MINUTES WERE USED PROCESSING DATA.
***
VOLUME FRACTION OF FISSILE MATERIAL IN THE CORE= 4.63033E-02

START TYPE 0 WAS USED.

THE NEUTRONS WERE STARTED WITH A FLAT DISTRIBUTION IN A CUBOID DEFINED BY:
+X= 1.24651E+02 -X=-1.75349E+02 +Y= 1.24651E+02 -Y=-1.75349E+02 +Z= 1.12016E+01 -Z= 0.00000E+00
THE FLAG TO START NEUTRONS IN THE REFLECTOR WAS TURNED OFF

0.43433 MINUTES WERE REQUIRED FOR STARTING. TOTAL ELAPSED TIME IS 0.44800 MINUTES.

```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

GENERATION	GENERATION K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
1	8.99752E-01	4.66000E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
2	9.15757E-01	4.85167E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132 WARNING...ONLY 975 INDEPENDENT FISSION POINTS WERE GENERATED						
3	8.81083E-01	5.08167E-01	8.81083E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	8.95401E-01	5.27333E-01	8.88242E-01	7.15911E-03	0.00000E+00	0.00000E+00
5	8.83993E-01	5.50167E-01	8.86825E-01	4.36926E-03	0.00000E+00	0.00000E+00
6	9.29745E-01	5.69500E-01	8.97555E-01	1.11658E-02	0.00000E+00	0.00000E+00
7	9.21681E-01	5.89667E-01	9.02380E-01	9.90387E-03	0.00000E+00	0.00000E+00
8	8.86503E-01	6.09667E-01	8.99734E-01	8.50845E-03	0.00000E+00	0.00000E+00
9	9.05096E-01	6.30833E-01	9.00500E-01	7.23163E-03	0.00000E+00	0.00000E+00
10	8.72586E-01	6.50000E-01	8.97011E-01	7.16920E-03	0.00000E+00	0.00000E+00
11	8.92029E-01	6.71000E-01	8.96457E-01	6.34683E-03	0.00000E+00	0.00000E+00
12	8.70002E-01	6.90333E-01	8.93812E-01	6.26294E-03	0.00000E+00	0.00000E+00
13	8.78336E-01	7.13167E-01	8.92405E-01	5.83712E-03	0.00000E+00	0.00000E+00
14	9.19176E-01	7.32333E-01	8.94636E-01	5.77669E-03	0.00000E+00	0.00000E+00
15	9.13891E-01	7.52500E-01	8.96117E-01	5.51635E-03	0.00000E+00	0.00000E+00
16	8.63450E-01	7.72667E-01	8.93784E-01	5.61493E-03	0.00000E+00	0.00000E+00
17	8.98462E-01	7.95500E-01	8.94096E-01	5.23651E-03	0.00000E+00	0.00000E+00
18	9.03556E-01	8.15667E-01	8.94687E-01	4.93387E-03	0.00000E+00	0.00000E+00
19	9.00677E-01	8.35833E-01	8.95039E-01	4.64794E-03	0.00000E+00	0.00000E+00
20	8.82546E-01	8.56833E-01	8.94345E-01	4.43674E-03	0.00000E+00	0.00000E+00
21	9.03549E-01	8.78000E-01	8.94830E-01	4.22460E-03	0.00000E+00	0.00000E+00
22	8.97150E-01	8.97167E-01	8.94946E-01	4.00949E-03	0.00000E+00	0.00000E+00
23	9.24417E-01	9.17333E-01	8.96349E-01	4.06379E-03	0.00000E+00	0.00000E+00
24	8.94524E-01	9.36500E-01	8.96266E-01	3.87556E-03	0.00000E+00	0.00000E+00
25	8.99957E-01	9.56667E-01	8.96426E-01	3.70671E-03	0.00000E+00	0.00000E+00
978	9.20387E-01	2.02703E+01	9.00271E-01	7.43267E-04	0.00000E+00	0.00000E+00
979	8.97272E-01	2.02905E+01	9.00268E-01	7.42512E-04	0.00000E+00	0.00000E+00
980	9.27792E-01	2.03107E+01	9.00296E-01	7.42286E-04	0.00000E+00	0.00000E+00
981	8.68171E-01	2.03298E+01	9.00263E-01	7.42253E-04	0.00000E+00	0.00000E+00
982	8.95645E-01	2.03508E+01	9.00258E-01	7.41510E-04	0.00000E+00	0.00000E+00
983	8.75421E-01	2.03710E+01	9.00233E-01	7.41186E-04	0.00000E+00	0.00000E+00
984	9.06928E-01	2.03902E+01	9.00240E-01	7.40463E-04	0.00000E+00	0.00000E+00
985	9.20851E-01	2.04103E+01	9.00261E-01	7.40006E-04	0.00000E+00	0.00000E+00
986	8.97710E-01	2.04315E+01	9.00258E-01	7.39258E-04	0.00000E+00	0.00000E+00
987	8.99171E-01	2.04515E+01	9.00257E-01	7.38508E-04	0.00000E+00	0.00000E+00
988	9.12688E-01	2.04717E+01	9.00270E-01	7.37867E-04	0.00000E+00	0.00000E+00
989	8.80411E-01	2.04918E+01	9.00250E-01	7.37393E-04	0.00000E+00	0.00000E+00
990	8.85791E-01	2.05120E+01	9.00235E-01	7.36792E-04	0.00000E+00	0.00000E+00
991	9.36212E-01	2.05322E+01	9.00271E-01	7.36945E-04	0.00000E+00	0.00000E+00
992	8.75343E-01	2.05513E+01	9.00246E-01	7.36631E-04	0.00000E+00	0.00000E+00
993	9.14852E-01	2.05705E+01	9.00261E-01	7.36034E-04	0.00000E+00	0.00000E+00
994	8.95530E-01	2.05917E+01	9.00256E-01	7.35308E-04	0.00000E+00	0.00000E+00
995	8.87804E-01	2.06118E+01	9.00244E-01	7.34674E-04	0.00000E+00	0.00000E+00
996	8.94301E-01	2.06310E+01	9.00238E-01	7.33959E-04	0.00000E+00	0.00000E+00
997	9.11712E-01	2.06520E+01	9.00249E-01	7.33311E-04	0.00000E+00	0.00000E+00
998	9.13794E-01	2.06722E+01	9.00263E-01	7.32701E-04	0.00000E+00	0.00000E+00
999	8.80324E-01	2.06923E+01	9.00243E-01	7.32239E-04	0.00000E+00	0.00000E+00
1000	8.92098E-01	2.07125E+01	9.00235E-01	7.31550E-04	0.00000E+00	0.00000E+00
1001	9.08588E-01	2.07345E+01	9.00243E-01	7.30865E-04	0.00000E+00	0.00000E+00
1002	8.77806E-01	2.07545E+01	9.00221E-01	7.30479E-04	0.00000E+00	0.00000E+00
1003	9.11293E-01	2.07747E+01	9.00232E-01	7.29833E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123

EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

LIFETIME = 4.13717E-05 + OR - 8.50918E-08 GENERATION TIME = 2.75056E-05 + OR - 4.33736E-08
 NU BAR = 2.44194E+00 + OR - 6.63172E-05 AVERAGE FISSION GROUP = 2.17289E+01 + OR - 3.82379E-03
 ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 2.78690E-01 + OR - 8.49711E-04

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.90025	+ OR - 0.00073	0.89952 TO 0.90098	0.89879 TO 0.90171	0.89806 TO 0.90244	1000000
4	0.90026	+ OR - 0.00073	0.89952 TO 0.90099	0.89879 TO 0.90172	0.89806 TO 0.90245	999000
5	0.90027	+ OR - 0.00073	0.89954 TO 0.90100	0.89881 TO 0.90174	0.89808 TO 0.90247	998000

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

6	0.90024	+ OR - 0.00073	0.89951 TO 0.90097	0.89878 TO 0.90171	0.89805 TO 0.90244	997000
7	0.90022	+ OR - 0.00073	0.89949 TO 0.90095	0.89876 TO 0.90169	0.89802 TO 0.90242	996000
8	0.90023	+ OR - 0.00073	0.89950 TO 0.90097	0.89877 TO 0.90170	0.89804 TO 0.90243	995000
9	0.90023	+ OR - 0.00073	0.89950 TO 0.90096	0.89876 TO 0.90170	0.89803 TO 0.90243	994000
10	0.90026	+ OR - 0.00073	0.89952 TO 0.90099	0.89879 TO 0.90172	0.89806 TO 0.90246	993000
11	0.90027	+ OR - 0.00073	0.89953 TO 0.90100	0.89880 TO 0.90173	0.89806 TO 0.90247	992000
12	0.90030	+ OR - 0.00073	0.89956 TO 0.90103	0.89883 TO 0.90177	0.89809 TO 0.90250	991000
17	0.90033	+ OR - 0.00074	0.89959 TO 0.90106	0.89885 TO 0.90180	0.89812 TO 0.90253	986000
22	0.90034	+ OR - 0.00074	0.89960 TO 0.90108	0.89886 TO 0.90182	0.89812 TO 0.90256	981000
937	0.90243	+ OR - 0.00297	0.89945 TO 0.90540	0.89648 TO 0.90837	0.89351 TO 0.91135	66000
942	0.90335	+ OR - 0.00304	0.90030 TO 0.90639	0.89726 TO 0.90944	0.89421 TO 0.91248	61000
947	0.90475	+ OR - 0.00314	0.90161 TO 0.90789	0.89846 TO 0.91103	0.89532 TO 0.91417	56000
952	0.90405	+ OR - 0.00320	0.90084 TO 0.90725	0.89764 TO 0.91046	0.89444 TO 0.91366	51000
957	0.90012	+ OR - 0.00284	0.89728 TO 0.90297	0.89444 TO 0.90581	0.89160 TO 0.90865	46000
962	0.90262	+ OR - 0.00288	0.89974 TO 0.90549	0.89687 TO 0.90837	0.89399 TO 0.91124	41000
967	0.90216	+ OR - 0.00306	0.89910 TO 0.90521	0.89604 TO 0.90827	0.89298 TO 0.91133	36000
972	0.90098	+ OR - 0.00343	0.89754 TO 0.90441	0.89411 TO 0.90785	0.89068 TO 0.91128	31000
977	0.89953	+ OR - 0.00348	0.89605 TO 0.90301	0.89257 TO 0.90650	0.88909 TO 0.90998	26000
982	0.89898	+ OR - 0.00366	0.89532 TO 0.90264	0.89166 TO 0.90630	0.88800 TO 0.90997	21000
987	0.89866	+ OR - 0.00434	0.89431 TO 0.90300	0.88997 TO 0.90735	0.88563 TO 0.91169	16000
992	0.89892	+ OR - 0.00413	0.89478 TO 0.90305	0.89065 TO 0.90719	0.88652 TO 0.91132	11000
997	0.89732	+ OR - 0.00656	0.89076 TO 0.90388	0.88420 TO 0.91043	0.87764 TO 0.91699	6000

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

PLOT OF AVERAGE K-EFFECTIVE BY GENERATION RUN.
THE LINE REPRESENTS K-EFF = 0.9003 + OR - 0.0007 WHICH OCCURS FOR 1003 GENERATIONS RUN.

0.8897 0.8987 0.9077

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

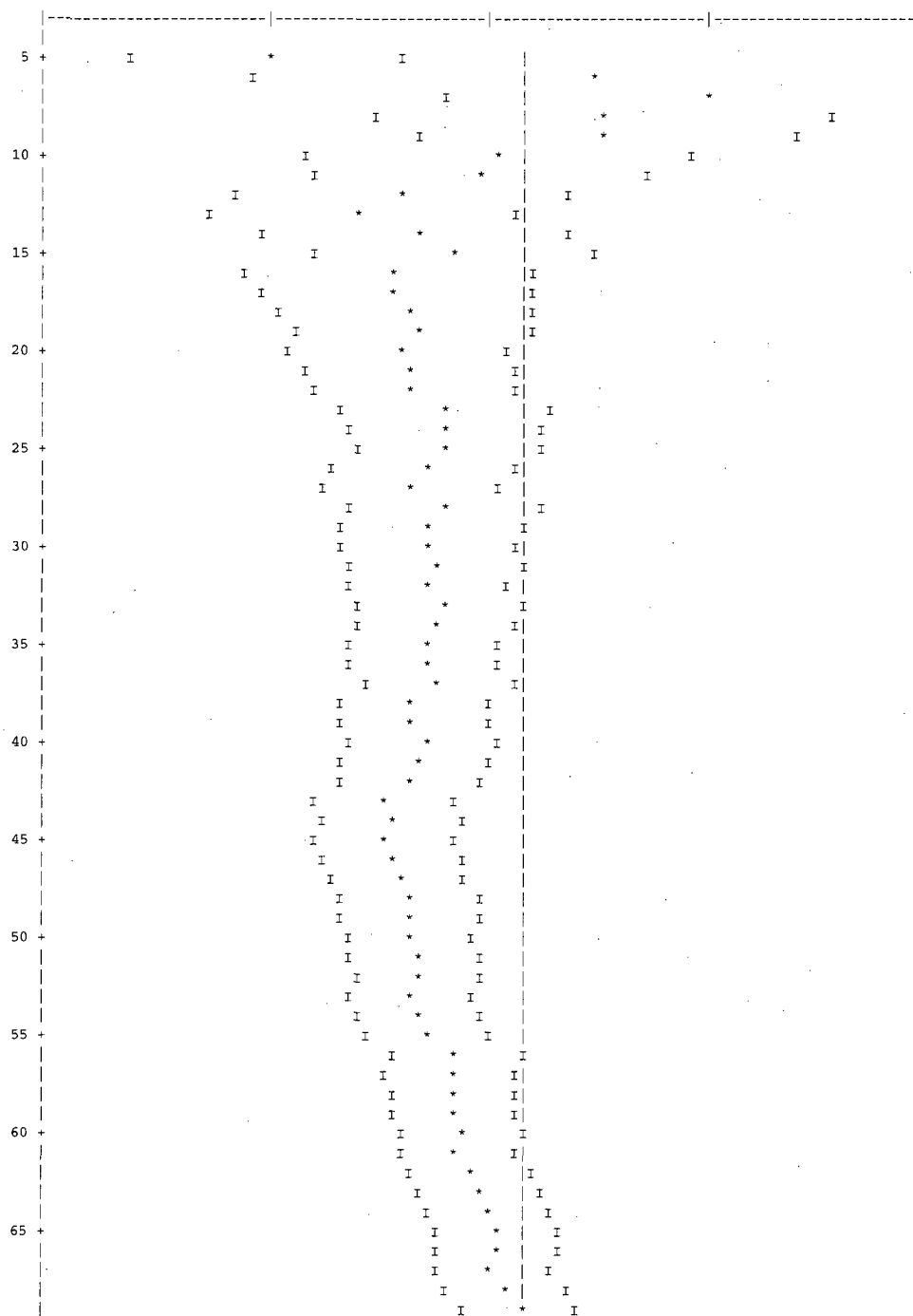


Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

965 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
970 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
975 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
980 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
985 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
990 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
995 +	I * I
	I * I
	I * I
	I * I
	I * I
	I * I
1000 +	I * I
	I * I
	I * I
	I * I
	I * I

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)									
SKIPPING 3 GENERATIONS									
GROUP	FISSION FRACTION	UNIT	REGION	FISSIONS	PERCENT DEVIATION	ABSORPTIONS	PERCENT DEVIATION	LEAKAGE	PERCENT DEVIATION
1	0.0059			5.34103E-03	1.0952	2.62006E-03	0.9146	0.00000E+00	0.0000
2	0.0233			2.09376E-02	0.3533	9.83986E-03	0.3143	0.00000E+00	0.0000
3	0.0254			2.28570E-02	0.3148	9.30230E-03	0.3096	0.00000E+00	0.0000
4	0.0105			9.47524E-03	0.3936	4.43581E-03	0.3867	0.00000E+00	0.0000
5	0.0034			3.05215E-03	0.2922	2.79613E-03	0.2823	0.00000E+00	0.0000
6	0.0031			2.76419E-03	0.2364	4.09041E-03	0.2261	0.00000E+00	0.0000
7	0.0030			2.73257E-03	0.2262	4.42282E-03	0.2137	0.00000E+00	0.0000
8	0.0031			2.75610E-03	0.2574	7.30715E-03	0.2466	0.00000E+00	0.0000
9	0.0042			3.78961E-03	0.2784	1.23266E-02	0.2587	0.00000E+00	0.0000
10	0.0089			8.03017E-03	0.2766	1.82729E-02	0.2692	0.00000E+00	0.0000
11	0.0185			1.66439E-02	0.2746	3.14249E-02	0.2662	0.00000E+00	0.0000
12	0.0235			2.11756E-02	0.3049	3.62575E-02	0.2950	0.00000E+00	0.0000
13	0.0217			1.94941E-02	0.3292	3.66911E-02	0.3201	0.00000E+00	0.0000
14	0.0168			1.51088E-02	0.3171	5.15006E-02	0.3029	0.00000E+00	0.0000
15	0.0037			3.32288E-03	0.5420	1.15041E-02	0.6567	0.00000E+00	0.0000
16	0.0025			2.28572E-03	0.6776	7.34526E-03	0.8560	0.00000E+00	0.0000
17	0.0039			3.48490E-03	0.8972	4.92116E-03	0.9181	0.00000E+00	0.0000
18	0.0051			4.57111E-03	0.9436	4.97165E-03	0.9175	0.00000E+00	0.0000
19	0.0063			5.68670E-03	0.7516	8.07060E-03	0.7823	0.00000E+00	0.0000
20	0.0264			2.37973E-02	0.4217	3.12588E-02	0.4003	0.00000E+00	0.0000
21	0.0142			1.28140E-02	0.6658	1.33961E-02	0.5770	0.00000E+00	0.0000
22	0.0333			3.00187E-02	0.4619	2.98115E-02	0.3983	0.00000E+00	0.0000
23	0.1108			9.97854E-02	0.2626	1.06269E-01	0.2026	0.00000E+00	0.0000
24	0.2010			1.80941E-01	0.1919	1.86480E-01	0.1344	0.00000E+00	0.0000
25	0.1650			1.48576E-01	0.2257	1.47079E-01	0.1524	0.00000E+00	0.0000
26	0.1936			1.74246E-01	0.2282	1.66781E-01	0.1555	0.00000E+00	0.0000
27	0.0628			5.65633E-02	0.4168	5.29893E-02	0.2909	0.00000E+00	0.0000
SYSTEM TOTAL =				9.00251E-01	0.0811	1.00216E+00	0.0205	0.00000E+00	0.0000

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

```

                                FREQUENCY FOR GENERATIONS    4 TO 1003
0.8183 TO 0.8213      *
0.8213 TO 0.8243
0.8243 TO 0.8273
0.8273 TO 0.8303      *
0.8303 TO 0.8333      *
0.8333 TO 0.8363
0.8363 TO 0.8392      *
0.8392 TO 0.8422      *
0.8422 TO 0.8452      *
0.8452 TO 0.8482      *
0.8482 TO 0.8512      *
0.8512 TO 0.8542      *
0.8542 TO 0.8572      *
0.8572 TO 0.8602      *
0.8602 TO 0.8632      *
0.8632 TO 0.8662      *
0.8662 TO 0.8691      *
0.8691 TO 0.8721      *
0.8721 TO 0.8751      *
0.8751 TO 0.8781      *
0.8781 TO 0.8811      *
0.8811 TO 0.8841      *
0.8841 TO 0.8871      *
0.8871 TO 0.8901      *
0.8901 TO 0.8931      *
0.8931 TO 0.8960      *
0.8960 TO 0.8990      *
0.8990 TO 0.9020      *
0.9020 TO 0.9050      *
0.9050 TO 0.9080      *
0.9080 TO 0.9110      *
0.9110 TO 0.9140      *
0.9140 TO 0.9170      *
0.9170 TO 0.9200      *
0.9200 TO 0.9229      *
0.9229 TO 0.9259      *
0.9259 TO 0.9289      *
0.9289 TO 0.9319      *
0.9319 TO 0.9349      *
0.9349 TO 0.9379      *
0.9379 TO 0.9409      *
0.9409 TO 0.9439      *
0.9439 TO 0.9469      *
0.9469 TO 0.9499      *
0.9499 TO 0.9528      *
0.9528 TO 0.9558      *
0.9558 TO 0.9588      *
0.9588 TO 0.9618      *
0.9618 TO 0.9648      *
0.9648 TO 0.9678      *
0.9678 TO 0.9708      *
```

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 254 TO 1003

0.8183 TO 0.8213	*
0.8213 TO 0.8243	
0.8243 TO 0.8273	
0.8273 TO 0.8303	.*
0.8303 TO 0.8333	*
0.8333 TO 0.8363	
0.8363 TO 0.8392	*
0.8392 TO 0.8422	
0.8422 TO 0.8452	*
0.8452 TO 0.8482	*
0.8482 TO 0.8512	*****
0.8512 TO 0.8542	*****
0.8542 TO 0.8572	*****
0.8572 TO 0.8602	*****
0.8602 TO 0.8632	*****
0.8632 TO 0.8662	*****
0.8662 TO 0.8691	*****
0.8691 TO 0.8721	*****
0.8721 TO 0.8751	*****
0.8751 TO 0.8781	*****
0.8781 TO 0.8811	*****
0.8811 TO 0.8841	*****
0.8841 TO 0.8871	*****
0.8871 TO 0.8901	*****
0.8901 TO 0.8931	*****
0.8931 TO 0.8960	*****
0.8960 TO 0.8990	*****
0.8990 TO 0.9020	*****
0.9020 TO 0.9050	*****
0.9050 TO 0.9080	*****
0.9080 TO 0.9110	*****
0.9110 TO 0.9140	*****
0.9140 TO 0.9170	*****
0.9170 TO 0.9200	*****
0.9200 TO 0.9229	*****
0.9229 TO 0.9259	*****
0.9259 TO 0.9289	*****
0.9289 TO 0.9319	*****
0.9319 TO 0.9349	*****
0.9349 TO 0.9379	*****
0.9379 TO 0.9409	*****
0.9409 TO 0.9439	*****
0.9439 TO 0.9469	*****
0.9469 TO 0.9499	****
0.9499 TO 0.9528	***
0.9528 TO 0.9558	***
0.9558 TO 0.9588	**
0.9588 TO 0.9618	*
0.9618 TO 0.9648	
0.9648 TO 0.9678	**
0.9678 TO 0.9708	*

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 504 TO 1003

0.8183 TO 0.8213	*
0.8213 TO 0.8243	
0.8243 TO 0.8273	
0.8273 TO 0.8303	*
0.8303 TO 0.8333	*
0.8333 TO 0.8363	
0.8363 TO 0.8392	
0.8392 TO 0.8422	
0.8422 TO 0.8452	*
0.8452 TO 0.8482	*
0.8482 TO 0.8512	*****
0.8512 TO 0.8542	***
0.8542 TO 0.8572	**
0.8572 TO 0.8602	***
0.8602 TO 0.8632	*****
0.8632 TO 0.8662	*****
0.8662 TO 0.8691	*****
0.8691 TO 0.8721	*****
0.8721 TO 0.8751	*****
0.8751 TO 0.8781	*****
0.8781 TO 0.8811	*****
0.8811 TO 0.8841	*****
0.8841 TO 0.8871	*****
0.8871 TO 0.8901	*****
0.8901 TO 0.8931	*****
0.8931 TO 0.8960	*****
0.8960 TO 0.8990	*****
0.8990 TO 0.9020	*****
0.9020 TO 0.9050	*****
0.9050 TO 0.9080	*****
0.9080 TO 0.9110	*****
0.9110 TO 0.9140	*****
0.9140 TO 0.9170	*****
0.9170 TO 0.9200	*****
0.9200 TO 0.9229	*****
0.9229 TO 0.9259	*****
0.9259 TO 0.9289	*****
0.9289 TO 0.9319	*****
0.9319 TO 0.9349	*****
0.9349 TO 0.9379	*****
0.9379 TO 0.9409	*****
0.9409 TO 0.9439	****
0.9439 TO 0.9469	***
0.9469 TO 0.9499	***
0.9499 TO 0.9528	*
0.9528 TO 0.9558	***
0.9558 TO 0.9588	**
0.9588 TO 0.9618	*
0.9618 TO 0.9648	
0.9648 TO 0.9678	**
0.9678 TO 0.9708	*

Figure 6.7-8 CSAS25 Input / Output for Enlarged Fuel Tube Model (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 754 TO 1003

```
0.8183 TO 0.8213 *
0.8213 TO 0.8243
0.8243 TO 0.8273
0.8273 TO 0.8303 *
0.8303 TO 0.8333 *
0.8333 TO 0.8363
0.8363 TO 0.8392
0.8392 TO 0.8422
0.8422 TO 0.8452
0.8452 TO 0.8482
0.8482 TO 0.8512 ****
0.8512 TO 0.8542 *
0.8542 TO 0.8572 *
0.8572 TO 0.8602
0.8602 TO 0.8632 ****
0.8632 TO 0.8662 *****
0.8662 TO 0.8691 *****
0.8691 TO 0.8721 ****
0.8721 TO 0.8751 ***
0.8751 TO 0.8781 ****
0.8781 TO 0.8811 *****
0.8811 TO 0.8841 *****
0.8841 TO 0.8871 *****
0.8871 TO 0.8901 *****
0.8901 TO 0.8931 *****
0.8931 TO 0.8960 *****
0.8960 TO 0.8990 *****
0.8990 TO 0.9020 *****
0.9020 TO 0.9050 *****
0.9050 TO 0.9080 *****
0.9080 TO 0.9110 *****
0.9110 TO 0.9140 *****
0.9140 TO 0.9170 *****
0.9170 TO 0.9200 *****
0.9200 TO 0.9229 *****
0.9229 TO 0.9259 *****
0.9259 TO 0.9289 *****
0.9289 TO 0.9319 *****
0.9319 TO 0.9349 *****
0.9349 TO 0.9379 *****
0.9379 TO 0.9409 *****
0.9409 TO 0.9439 **
0.9439 TO 0.9469
0.9469 TO 0.9499 *
0.9499 TO 0.9528 *
0.9528 TO 0.9558 *
0.9558 TO 0.9588 **
0.9588 TO 0.9618 *
0.9618 TO 0.9648
0.9648 TO 0.9678 *
0.9678 TO 0.9708 *
```

CONGRATULATIONS! YOU HAVE SUCCESSFULLY TRAVERSED THE PERILOUS PATH THROUGH KENO V IN 20.77833 MINUTES

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition

1

```

AA      N      NN      SSSSSS  WW      WW      EEEEEEEE  RRRRRRRR  SSSSSS      CCCCCC  000000  DDDDDDD  EEEEEEEE
AAAA    NN      NN      SSSSSSSS  WW      WW      EEEEEEEE  RRRRRRRR  SSSSSSSS  CCCCCCCC  00000000  DDDDDDDD  EEEEEEEE
AA  AA  NNN      NN      SS      WW      WW      EE      RR      RR      SS      CC      00      00      DD      DD      EE
AA  AA  NNNN      NN      SS      WW      WW      EE      RR      RR      SS      CC      00      00      DD      DD      EE
AAAAAAA  NN      NN      NN      SSSSSS  WW      WW      EEEEE  RRRRRRRR  SSSSSS      CC      00      00      DD      DD      EEEEE
AAAAAAA  NN      NN      NN      SSSSSS  WW      WW      EEEEE  RRRRRRRR  SSSSSS      CC      00      00      DD      DD      EEEEE
AA      AA  NN      NNNN      SS      WW      WW      WW      EE      RR      RR      SS      CC      00      00      DD      DD      EE
AA      AA  NN      NNN      SS      WW      WW      WW      EE      RR      RR      SS      CC      00      00      DD      DD      EE
AA      AA  NN      NN      SSSSSSSS  WWWWWW  EEEEEEEE  RR      RR      SSSSSSSS  CCCCCCCC  00000000  DDDDDDDD  EEEEEEEE
AA      AA  NN      N      SSSSSS      WW      WW      EEEEEEEE  RR      RR      SSSSSS      CCCCCC      000000      DDDDDDD      EEEEEEEE

      M      M      000000  N      NN      KK      KK      888888      AA
MM      MM      00000000  NN      NN      KK      KK      88888888      AAAA
MMM      MM      00      00      NNN      NN      KK      KK      88      88      AA      AA
MMMMMMMM      00      00      NNN      NN      KK      KK      88      88      AA      AA
MM      MM      MM      00      00      NN      NN      NN      KKKK      888888      AAAAAAAA
MM      MM      MM      00      00      NN      NN      NN      KKKK      888888      AAAAAAAA
MM      MM      MM      00      00      NN      NNNN      KK      KK      88      88      AA      AA
MM      MM      MM      00      00      NN      NN      NN      KK      KK      88      88      AA      AA
MM      MM      00000000  NN      NN      KK      KK      88888888      AA      AA
MM      MM      000000      NN      N      KK      KK      888888      AA      AA

```

Running on machine 57NST (Windows NT)
Date and time of execution: 15/ 9/2000 - 17.34.12

```

PPPPPPP  CCCCCC      WW      WW      IIIIII  N      NN      N      NN      TTTTTTT
PPPPPPPP CCCCCCCC      WW      WW      IIIIII  NN      NN      NN      NN      TTTTTTT
PP      PP      CC      WW      WW      II      NNN      NN      NNN      NN      TT
PP      PP      CC      WW      WW      II      NNNN      NN      NNNN      NN      TT
PPPPPPPP CC      WW      WW      II      NN      NN      NN      NN      NN      NN      TT
PPPPPPPP CC      WW      WW      II      NN      NN      NN      NN      NN      NN      TT
PP      CC      WW      WW      WW      II      NN      NNNN      NN      NNNN      TT
PP      CC      WW      WW      WW      II      NN      NNN      NN      NNN      TT
PP      CCCCCC      WWWWWW  IIIIII  NN      NN      NN      NN      TT
PP      CCCCCC      WW      WW      IIIIII  NN      N      NN      N      TT

```

***** This Computer Program is Supplied Under Licence by the AEA Technology ANSWERS Software Service *****

```

*****
*      Program MONK 8A - Release Update 1      *
*      -----      *
*      *
*      This is the ANSWERS QA Set version of MONK. This      *
*      program has successfully executed the designated set      *
*      of test cases on the ANSWERS QA Set computer systems.      *
*      *
*      This is the first update release of MONK8A, known as      *
*      the RUI release. It contains corrections to the errors      *
*      reported in ANSWERS/CRIT/ERROR(98)28,30,31 and 33      *
*      *
*      22 January 1999      *
*      *
*      The MONK program is developed and maintained      *
*      through a collaboration between AEA Technology PLC      *
*      and British Nuclear Fuels PLC.      *
*      *
*****

```

```

*****
***      ANSWERS Software Licensing Procedure      ***
*****

*** The timelock password has been successfully verified.
*** This password expires at midnight on 30/ 6/2001

*** Successful authorisation achieved on system 57NST
*** This system has the identification number 131008778

*****
This is a licensed ANSWERS software product made available to
NAC International.
The use of this program is restricted within the terms of your
licence agreement with AEA Technology.
This copy is registered with version identifier - PCNT9
*****

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
*****
***      End of ANSWERS Software Licensing Checks      ***
*****

1PRINT OF INPUT DATA SET
=====
*
* Connecticut Yankee STC Transfer Cask Model *
*
always convert
! Always converts intergers to real numbers

*****

*****
* Inserting TSC Parameters Here *
*
*****

*****
< FPF LABEL FUELPARAM
< CPF LABEL TUBEPARAM
< CPF LABEL BASKPARAM
< CPF LABEL CANPARAM
*****

* The following parameters need to be updated

*****
@tscMAT  = 15                ! Last material number of TSC going into cask
@tscNUM  = 20                ! Last part number of TSC going into cask
@fuelMAT = 1                 ! Material number of fuel from TSC model
*****

* The remaining parameters should stay the same unless the drawings change.

@cpi      = 2.54              ! Centimeters per inch
@tschT    = 151.75            ! Height of TSC going into cask = 151.75*2.54
@tscoD    = 70.64             ! Outer diameter of TSC going into cask = 70.64*2.54

@tfrOD    = 89.0              ! Transfer cask OD = 89.0*2.54

@cavMAT   = 4                 ! Cavity Material additive

@btmplHT  = 1                 ! Bottom plate height = 1.0*2.54
@btmplMAT = 1                 ! Bottom plate material additive

@inshlHT  = 150.0             ! Inner shell height = 149.8*2.54
@inshlID  = 71.5              ! Inner shell inner Diameter = 71.5*2.54
@inshlTH  = 0.75              ! Inner shell thickness = 0.75*2.54
@inshlMAT = 1                 ! Inner shell material additive

@pbshlTH  = 4.0               ! Lead shell thickness = 4.0*2.54
@pbshlMAT = 2                 ! Lead shell material additive

@ns4frGAP = 4                 ! Gap above Lead filled w/ NS-4-FR = 3*2.54
@n4shlMAT = 3                 ! NS-4-FR shell material additive

@otshlTH  = 1.25              ! Outer shell thickness = 1.25*2.54
@otshlMAT = 1                 ! Outer shell material additive

@topplHT  = 2.0               ! Top plate height = 2.0*2.54
@topplMAT = 1                 ! Top plate material additive

@rtarnHT  = 0.75              ! Retaining ring height = 0.75*2.54
@rtarnID  = 68.5              ! Retaining ring inner diameter = 68.5*2.54
@rtarnOD  = 80.8              ! Retaining ring outer diameter = 80.8*2.54
@rtarnMAT = 1                 ! Retaining ring material additive

@shldrHT  = 9.50              ! Shield door height = 9.50*2.54
@shldrMAT = 1                 ! Shield door material additive
@drailID  = 75.5              ! Shield door rail inner dimension = 75.5*2.54
@drailWD  = 4.5               ! Shield door rail width = 4.5
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

```
@trunzcor = 15.00      ! Distance b/w origen of trunion cylinder and retaining ri
@trunOD  = 10.00      ! Diameter of trunions = 10*2.54
@trunMAT = 1          ! Trunion material additive

*****
BEGIN MATERIAL SPECIFICATION
*****

NORMALISE

*****
< FMF LABEL FUELMIX
*****

NMATERIALS [@tscMAT + 5]

*****
* Inserting TSC Materials Here *
*
< FMF LABEL FUELMAT
< CMF LABEL TUBEMAT
< CMF LABEL BASKETMAT
< CMF LABEL CANMAT
*****

WEIGHT
MATERIAL [@tscMAT + 1] STAINLESS 304L STEEL      !Steel components of Transfer Cask

WEIGHT
MATERIAL [@tscMAT + 2] DENSITY 11.04             !LEAD
PB      PROP 1.0000

ATOMS
MATERIAL [@tscMAT + 3] DENSITY 0.0               !NS-4-FR
B10     PROP 8.553E-5
B11     PROP 3.422E-4
AL27    PROP 7.796E-3
H1      PROP 5.854E-2
O16     PROP 2.609E-2
C       PROP 2.264E-2
N       PROP 1.394E-3

ATOMS
MATERIAL [@tscMAT + 4] DENSITY 0.9982            ! Material (water) outside of Basket
H1      PROP 2
O16     PROP 1

ATOMS
MATERIAL [@tscMAT + 5] DENSITY 0.9982            ! Additional water for future use
H1      PROP 2
O16     PROP 1
END

*****
BEGIN MATERIAL GEOMETRY
*****

*****
* Inserting TSC Geometry Here *
*
< cgf Label BASKGEOM
*****

!234567890123456789012345678901234567890123456789012345678901234567890123456789012
PART [@tscNUM + 1]
ZROD 1 3*0  [@cpi*@tscOD/2]  [@cpi*@tscHT]      ! TSC
ZROD 2 3*0  [@cpi*@inshlID/2]  [@cpi*(@btmplHT+@inshlHT+@topplHT)] ! Cask cavity
ZROD 3 3*0  [@cpi*@tfrOD/2]  [@cpi*@btmplHT]    ! Bottom plate
ZROD 4 2*0  [@cpi*@btmplHT]  [@cpi*(@inshlID/2)+@inshlTH)]
                                     !
                                     Inner      shell
ZROD 5 2*0  [@cpi*@btmplHT]  [@cpi*(@inshlID/2)+@inshlTH+@pbshlTH)]
                                     !
                                     ! Lead shell
ZROD 6 2*0  [@cpi*@btmplHT]  [@cpi*(@tfrOD/2)-@otshlTH)]
                                     !
                                     NS-4-FR      shell
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

```
ZROD 7 2*0  [ecpi*@btmplHT] [ecpi*(@tfrOD/2)] [ecpi*@inshlHT]           ! Outer shell
ZROD 8 2*0  [ecpi*(@btmplHT+@inshlHT)] [ecpi*(@tfrOD/2)]
               [ecpi*@topplHT]           ! Top plate
ZROD 9 2*0  [ecpi*(@btmplHT+@inshlHT+@topplHT)] [ecpi*@rtnrnID/2]
               [ecpi*@rtnrnHT]           ! Area inside retaining ring
ZROD 10 2*0 [ecpi*(@btmplHT+@inshlHT+@topplHT)] [ecpi*@rtnrnOD/2]
               [ecpi*@rtnrnHT]           ! Retaining ring
ZROD 11 2*0 [-@shldrHT*ecpi] [ecpi*@tfrOD/2] [ecpi*@shldrHT]           ! Shield doors and
YP 12  [ecpi*(@drrailID/2)+@drrailWD)]           ! Y plane for shield door rail cu
YP 13  [-@ecpi*(@drrailID/2)+@drrailWD)]           ! Y plane for shield door rail c
XROD 14  [-@ecpi*@tfrOD/2] 0
               [ecpi*(@btmplHT+@inshlHT+@topplHT - @trunzcor)]
               [ecpi*@trunOD/2] [ecpi*@tfrOD]           ! Trunions
ZROD 15 2*0 [-1 * @ecpi*@shldrHT] [ecpi*@tfrOD/2]
               [ecpi*(@shldrHT+@btmplHT+@inshlHT+@topplHT+@rtnrnHT)] ! Container
```

ZONES

```
P@tscNUM           +1           ! TSC
M[@tscMAT + @cavMAT] +2 -1       ! Cask cavity
M[@tscMAT + @btmplMAT] +3 -2     ! Bottom plate
M[@tscMAT + @inshlMAT] +4 -2 -14 ! Inner shell
M[@tscMAT + @pbshlMAT] +5 -4 -14 ! Lead shell
M[@tscMAT + @n4shlMAT] +6 -5 -4 -14 ! NS-4-FR shell
M[@tscMAT + @otshlMAT] +7 -6 -14 ! Outer shell
M[@tscMAT + @topplMAT] +8 -2     ! Top plate
M[@tscMAT + @cavMAT] +9         ! Area inside retaining ring
M[@tscMAT + @rtnrnMAT] +10 -9    ! Retaining ring
M[@tscMAT + @shldrMAT] +11 -12 +13 ! Shield doors and rails
M[@tscMAT + @cavMAT] +11 +12     ! Space outside of shield door
M[@tscMAT + @cavMAT] +11 -13     ! Space outside of shield door
M[@tscMAT + @trunMAT] +14 +15 -2 ! Trunions
M[@tscMAT + @cavMAT] +15 -14 -11 -10 -8 -7 -3 ! Container of same material f
```

ALBEDO 1 1 1
PERIODIC

END

BEGIN HOLE DATA

* Inserting TSC Hole Data Here *

*

< FHF LABEL FUELHOLE

< CHF LABEL TUBEHOLE

< CHF LABEL BASKHOLE

END

BEGIN CONTROL DATA

*READ ! read and check each independently

*SEEK MULTIPLE DEFINITIONS

SEEDS 54321 54321

STAGES -3 810 1000 STDV 0.0008

END

BEGIN SOURCE GEOMETRY

ZONEMAT

ALL / MATERIAL @fuelMAT

END

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```

Changing input to named channel:      FPF after reading
15 lines from named channel:      INPUT Total line count= 15
Real parameter number 1:rpitch      = 1.442700E+00
Real parameter number 2:lactfu      = 3.061970E+02
Real parameter number 3:odpellet    = 9.307000E-01
Real parameter number 4:odclad      = 1.066800E+00
Real parameter number 5:cladth      = 5.660000E-02
Real parameter number 6:odgt        = 1.346200E+00
Real parameter number 7:gtth        = 3.300000E-02
Real parameter number 8:odit        = 1.346200E+00
Real parameter number 9:itth        = 3.300000E-02
Real parameter number 10:lfu        = 3.218688E+02
Real parameter number 11:lrcap      = 1.739900E+00
Real parameter number 12:lbcap      = 1.739900E+00
Real parameter number 13:lfexpb     = 0.000000E+00
Real parameter number 14:lattice    = 1.500000E+01
Real parameter number 15:lassem     = 3.481324E+02
Real parameter number 16:wfuel      = 2.164080E+01
Real parameter number 17:lbnozz     = 8.097500E+00
Real parameter number 18:ltnozz     = 1.724660E+01
Real parameter number 19:fgapb      = 0.000000E+00
Real parameter number 20:fuoffz     = 0.000000E+00
Returning input to named channel:      INPUT after reading
33 lines from named channel:      FPF Total line count= 45

Changing input to named channel:      CPF after reading
16 lines from named channel:      INPUT Total line count= 46
Real parameter number 21:wft        = 2.222500E+01
Real parameter number 22:ftth       = 1.219000E-01
Real parameter number 23:lfutube    = 3.357880E+02
Real parameter number 24:wfto       = 2.294130E+01
Integer parameter number 25:ftoffz   = 0
Real parameter number 26:asoffx     = 5.842000E-01
Real parameter number 27:asoffy     = 5.842000E-01
Real parameter number 28:asoffyp    = 2.921000E-01
Real parameter number 29:asoffxo    = 1.600200E+00
Real parameter number 30:asoffyo    = 1.600200E+00
Real parameter number 31:wbrl       = 2.025400E+01
Real parameter number 32:blt        = 1.905000E-01
Real parameter number 33:blct       = 1.270000E-01
Real parameter number 34:lboral     = 3.261360E+02
Real parameter number 35:boffz      = 3.556000E-01
Integer parameter number 36:bshiftx  = 0
Integer parameter number 37:bshifty  = 0
Integer parameter number 38:bshifty  = 0
Real parameter number 39:cvst       = 4.570000E-02
Real parameter number 40:lcvs       = 3.284220E+02
Real parameter number 41:cvoffz     = 2.032000E+00
Real parameter number 42:tboffx     = 3.886000E-01
Real parameter number 43:tboffy     = 3.886000E-01
Real parameter number 44:tboffyp    = 1.943000E-01
Real parameter number 45:overs      = 1.016000E+00
Returning input to named channel:      INPUT after reading
47 lines from named channel:      CPF Total line count= 87

Changing input to named channel:      CPF after reading
17 lines from named channel:      INPUT Total line count= 88
Real parameter number 46:diabw      = 1.752092E+02
Real parameter number 47:lbw        = 5.080000E+00
Real parameter number 48:lbwd       = 1.270000E+00
Real parameter number 49:lbws       = 1.038860E+01
Real parameter number 50:diatpw     = 1.752092E+02
Real parameter number 51:ltpw       = 1.727200E+01
Real parameter number 52:ltpwd      = 1.270000E+00
Real parameter number 53:lspring    = 1.473200E+01
Real parameter number 54:thspring    = 1.270000E+00
Real parameter number 55:opspd      = 2.332990E+01
Real parameter number 56:diassp     = 1.756410E+02
Real parameter number 57:tspd       = 1.270000E+00
Real parameter number 58:wsspx1     = 1.353330E+01
Real parameter number 59:wsspx2     = 3.961910E+01
Real parameter number 60:wsspx3     = 6.545090E+01
Real parameter number 61:tsspy1     = 2.708420E+01
Real parameter number 62:tsspy2     = 5.926600E+01

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

```
Real parameter number 63:wsspxo = 4.012710E+01
Real parameter number 64:tsspyo = 5.405900E+01
Real parameter number 65:diaht = 1.749298E+02
Real parameter number 66:thtd = 1.270000E+00
Integer parameter number 67:nhtd = 27
Real parameter number 68:dsspht = 4.559300E+00
Real parameter number 69:lbasket = 3.594100E+02
Real parameter number 70:diskstack = 3.160522E+02
Integer parameter number 71:baskoff = 0
Returning input to named channel: INPUT after reading
96 lines from named channel: CPF Total line count= 133
```

```
Changing input to named channel: CPF after reading
18 lines from named channel: INPUT Total line count= 134
Real parameter number 72:canod = 1.794256E+02
Real parameter number 73:canth = 1.587500E+00
Real parameter number 74:canl = 3.854450E+02
Real parameter number 75:canbot = 4.445000E+00
Real parameter number 76:shieldl = 1.270000E+01
Real parameter number 77:structl = 7.620000E+00
Real parameter number 78:cavheight = 3.606800E+02
```

```
Returning input to named channel: INPUT after reading
113 lines from named channel: CPF Total line count= 147
Integer parameter number 79:tscMAT = 15
Integer parameter number 80:tscNUM = 20
Integer parameter number 81:fuelMAT = 1
Real parameter number 82:cpi = 2.540000E+00
Real parameter number 83:tscht = 1.517500E+02
Real parameter number 84:tscOD = 7.064000E+01
Real parameter number 85:tfrOD = 8.900000E+01
Integer parameter number 86:cavMAT = 4
Integer parameter number 87:btmplHT = 1
Integer parameter number 88:btmplMAT = 1
Real parameter number 89:inshlHT = 1.500000E+02
Real parameter number 90:inshlID = 7.150000E+01
Real parameter number 91:inshlTH = 7.500000E-01
Integer parameter number 92:inshlMAT = 1
Real parameter number 93:pbshlTH = 4.000000E+00
Integer parameter number 94:pbshlMAT = 2
Integer parameter number 95:ns4frGAP = 4
Integer parameter number 96:n4shlMAT = 3
Real parameter number 97:otshlTH = 1.250000E+00
Integer parameter number 98:otshlMAT = 1
Real parameter number 99:topplHT = 2.000000E+00
Integer parameter number 100:topplMAT = 1
Real parameter number 101:rttrnHT = 7.500000E-01
Real parameter number 102:rttrnID = 6.850000E+01
Real parameter number 103:rttrnOD = 8.080000E+01
Integer parameter number 104:rttrnMAT = 1
Real parameter number 105:shldrHT = 9.500000E+00
Integer parameter number 106:shldrMAT = 1
Real parameter number 107:drrailID = 7.550000E+01
Real parameter number 108:drrailWD = 4.500000E+00
Real parameter number 109:trunzcor = 1.500000E+01
Real parameter number 110:trunOD = 1.000000E+01
Integer parameter number 111:trunMAT = 1
```

```
*****
* ENTERING UNIT 9 TO READ MATERIAL SPECIFICATION *
*****
```

Free format reading told to echo all input lines to channel 6

```
NORMALISE
+++++FILE NAME KEY IS DICE96J2V5.DAT
DATABASE NAME IS d:\answers\data_libraries\monk_matdbv2d.dat
DATABASE HISTORY IS
MONK MATERIAL SPECIFICATION MODULE DATABASE VERSION 2C, STRUCTURING :
. DICE96J2V4.DAT
. DICE96J2V5.DAT
. OILDICE96V2.DAT
. DICE97E6V1.DAT
. DICE97E6V2.DAT
. DICE97E6V3.DAT
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
. DICE96J2V5.DAT .
. DICE97E6V3.DAT
*****
* FOR USE WITH MONK, NOT FOR MCBEND AND RANKERN *
*
* If a requested nuclide is not present in the *
* nuclear data library, a warning message will be *
* printed below and the nuclide will be omitted *
* from the material. *
* CHECK CAREFULLY that this automatic correction *
* to the material composition is appropriate. *
*****

< FMF LABEL FUELMIX
Changing input to named channel:      FMF after reading
80 lines from named channel:      INPUT Total line count= 209

NMIXTURES 1
*****

ATOMS
MIXTURE 1
H1 2
O16 1

< UP
Returning input to named channel:      INPUT after reading
10 lines from named channel:      FMF Total line count= 218

NMATERIALS [@tscMAT + 5]
*****
The input characters      : [@tscMAT + 5]
have been replaced by    : +20.00000

< FMF LABEL FUELMAT

Changing input to named channel:      FMF after reading
88 lines from named channel:      INPUT Total line count= 226

WEIGHT
MATERIAL 1 DENSITY 10.412      ! 95% UO2 theoretical density 4.2 wt% 235U
*****
* Inserting TSC Materials Here *
*
U235 PROP 0.04063
U238 PROP 0.84080
O16 PROP 0.11856

WEIGHT
MATERIAL 2 ZIRCONIUM          ! Fuel pin cladding / End Caps
++++MATERIAL < 2>: USES MIXTURE <ZIRCONIUM>

ATOMS
++++MATERIAL < 2>: DEFAULT PROPORTION=1.0
MATERIAL 3 DENSITY 0.9982      ! Water In Lattice and Tube
H1 PROP 2
O16 PROP 1

ATOMS
MATERIAL 4 DENSITY 0.9982      ! Water In Fuel Rod Clad Gap
H1 PROP 2
O16 PROP 1

WEIGHT
MATERIAL 5 ZIRCONIUM          ! Guide tube material
++++MATERIAL < 5>: USES MIXTURE <ZIRCONIUM>

WEIGHT
++++MATERIAL < 5>: DEFAULT PROPORTION=1.0
MATERIAL 6 ZIRCONIUM          ! Instrument Tube Material
++++MATERIAL < 6>: USES MIXTURE <ZIRCONIUM>

VOLUME
++++MATERIAL < 6>: DEFAULT PROPORTION=1.0
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
MATERIAL 7
STAINLESS 304L STEEL PROP 0.3031
++++MATERIAL < 7>: USES MIXTURE <STAINLESS 304L STEEL>
MIXTURE 1 DENSITY 0.9982 PROP 0.6969 ! Lower Nozzle Material

MATERIAL 8
STAINLESS 304L STEEL PROP 0.1720
++++MATERIAL < 8>: USES MIXTURE <STAINLESS 304L STEEL>
MIXTURE 1 DENSITY 0.9982 PROP 0.8280 ! Upper Nozzle Material

< UP
Returning input to named channel: INPUT after reading
48 lines from named channel: FMF Total line count= 262
< CMF LABEL TUBEMAT

Changing input to named channel: CMF after reading
89 lines from named channel: INPUT Total line count= 263

WEIGHT
MATERIAL 9 STAINLESS 304L STEEL ! Tube wall and cover sheet
++++MATERIAL < 9>: USES MIXTURE <STAINLESS 304L STEEL>

VOLUME
++++MATERIAL < 9>: DEFAULT PROPORTION=1.0
MATERIAL 10 ! BORAL core
AL27 DENSITY 2.6226 PROP 0.5738
B10 DENSITY 2.6226 PROP 0.0450
B11 DENSITY 2.6226 PROP 0.2735
C DENSITY 2.6226 PROP 0.0926
VOID PROP 0.0150

WEIGHT
MATERIAL 11 ALUMINIUM ! BORAL aluminum clad

< UP
Returning input to named channel: INPUT after reading
24 lines from named channel: CMF Total line count= 279
< CMF LABEL BASKETMAT

Changing input to named channel: CMF after reading
90 lines from named channel: INPUT Total line count= 280
WEIGHT
++++MATERIAL < 11>: USES MIXTURE <ALUMINIUM>
++++MATERIAL < 11>: DEFAULT PROPORTION=1.0
MATERIAL 12 STAINLESS 304L STEEL ! Structural Disk Material
++++MATERIAL < 12>: USES MIXTURE <STAINLESS 304L STEEL>

WEIGHT
++++MATERIAL < 12>: DEFAULT PROPORTION=1.0
MATERIAL 13 STAINLESS 304L STEEL ! Weldment Material
++++MATERIAL < 13>: USES MIXTURE <STAINLESS 304L STEEL>

WEIGHT
++++MATERIAL < 13>: DEFAULT PROPORTION=1.0
MATERIAL 14 ALUMINIUM ! Heat Transfer Disk Material

< UP
Returning input to named channel: INPUT after reading
41 lines from named channel: CMF Total line count= 291
< CMF LABEL CANMAT

Changing input to named channel: CMF after reading
91 lines from named channel: INPUT Total line count= 292
WEIGHT
++++MATERIAL < 14>: USES MIXTURE <ALUMINIUM>
++++MATERIAL < 14>: DEFAULT PROPORTION=1.0
MATERIAL 15 STAINLESS 304L STEEL ! Canister Material
++++MATERIAL < 15>: USES MIXTURE <STAINLESS 304L STEEL>

< UP
Returning input to named channel: INPUT after reading
52 lines from named channel: CMF Total line count= 297
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

```

WEIGHT
      *****MATERIAL < 15>: DEFAULT PROPORTION=1.0
MATERIAL [@tscMAT + 1] STAINLESS 304L STEEL      !Steel components of Transfer Cask
*****
The input characters      : [@tscMAT + 1]
have been replaced by    : -16.00000
      *****MATERIAL < 16>: USES MIXTURE <STAINLESS 304L STEEL>

WEIGHT
      *****MATERIAL < 16>: DEFAULT PROPORTION=1.0
MATERIAL [@tscMAT + 2] DENSITY 11.04             !LEAD
The input characters      : [@tscMAT + 2]
have been replaced by    : +17.00000
PB      PROP 1.0000

ATOMS
MATERIAL [@tscMAT + 3] DENSITY 0.0               !NS-4-FR
The input characters      : [@tscMAT + 3]
have been replaced by    : +18.00000
B10     PROP 8.553E-5
B11     PROP 3.422E-4
AL27    PROP 7.796E-3
H1      PROP 5.854E-2
O16     PROP 2.609E-2
C       PROP 2.264E-2
N       PROP 1.394E-3

ATOMS
MATERIAL [@tscMAT + 4] DENSITY 0.9982            ! Material (water) outside of Basket
The input characters      : [@tscMAT + 4]
have been replaced by    : +19.00000
H1      PROP 2
O16     PROP 1

ATOMS
MATERIAL [@tscMAT + 5] DENSITY 0.9982            ! Additional water for future use
The input characters      : [@tscMAT + 5]
have been replaced by    : +20.00000
H1      PROP 2
O16     PROP 1
END

=====
MATERIAL #  LIBRARY  DFN  MOULD  ALIAS      ATOM/B-CM      NADLIB  DENSITY  MASS  Z.A
           SUBSTANCE
MATERIAL  1  U235    9228   62   J2U235    1.08389E-03   U235  1.04120E+01  4.06304E-02  92.235
           1  U238    9237   63   J2U238    2.21469E-02   U238  1.04120E+01  8.40808E-01  92.238
           1  O16     825    33   J2O16    4.64778E-02   O  1.04120E+01  1.18561E-01  8.016
MATERIAL  2  ZR     4000   47   J2ZR     4.29096E-02   ZR  6.50000E+00  1.00000E+00  40.000
MATERIAL  3  H1     10293  64   J2HINH2O  6.67530E-02   H  9.98200E-01  1.11915E-01  1.001
           3  O16     825    33   J2O16    3.33765E-02   O  9.98200E-01  8.88085E-01  8.016
MATERIAL  4  H1     10293  64   J2HINH2O  6.67530E-02   H  9.98200E-01  1.11915E-01  1.001
           4  O16     825    33   J2O16    3.33765E-02   O  9.98200E-01  8.88085E-01  8.016
MATERIAL  5  ZR     4000   47   J2ZR     4.29096E-02   ZR  6.50000E+00  1.00000E+00  40.000
MATERIAL  6  ZR     4000   47   J2ZR     4.29096E-02   ZR  6.50000E+00  1.00000E+00  40.000
MATERIAL  7  FE54    2625   15   J2FE54    1.13160E-03   FE  3.09913E+00  3.27048E-02  26.054
           7  FE56    2631   66   J2FE56    1.75916E-02   FE  3.09913E+00  5.27228E-01  26.056
           7  FE57    2634   16   J2FE57    4.02773E-04   FE  3.09913E+00  1.22872E-02  26.057
           7  FE58    2637   17   J2FE58    5.37031E-05   FE  3.09913E+00  1.66701E-03  26.058
           7  CR50    2425   11   J2CR50    2.17721E-04   CR  3.09913E+00  5.82656E-03  24.050
           7  CR52    2431   53   J2CR52    4.19859E-03   CR  3.09913E+00  1.16848E-01  24.052
           7  CR53    2434   12   J2CR53    4.76031E-04   CR  3.09913E+00  1.35031E-02  24.053
           7  CR54    2437   13   J2CR54    1.18507E-04   CR  3.09913E+00  3.42495E-03  24.054
           7  NI58    2825   56   J2NI58    1.34311E-03   NI  3.09913E+00  4.16931E-02  28.058
           7  NI60    2831   57   J2NI60    5.17360E-04   NI  3.09913E+00  1.66132E-02  28.060
           7  NI61    2834   58   J2NI61    2.24913E-05   NI  3.09913E+00  7.34284E-04  28.061
           7  NI62    2837   59   J2NI62    7.16961E-05   NI  3.09913E+00  2.37900E-03  28.062
           7  NI64    2843   60   J2NI64    1.82693E-05   NI  3.09913E+00  6.25781E-04  28.064
           7  H1     10293  64   J2HINH2O  4.65202E-02   H  3.09913E+00  2.51210E-02  1.001
           7  O16     825    33   J2O16    2.32601E-02   O  3.09913E+00  1.99344E-01  8.016
MATERIAL  8  FE54    2625   15   J2FE54    6.42149E-04   FE  2.19041E+00  2.62583E-02  26.054
           8  FE56    2631   66   J2FE56    9.98270E-03   FE  2.19041E+00  4.23306E-01  26.056
           8  FE57    2634   16   J2FE57    2.28562E-04   FE  2.19041E+00  9.86528E-03  26.057
           8  FE58    2637   17   J2FE58    3.04749E-05   FE  2.19041E+00  1.33842E-03  26.058
           8  CR50    2425   11   J2CR50    1.23550E-04   CR  2.19041E+00  4.67809E-03  24.050
           8  CR52    2431   53   J2CR52    2.38257E-03   CR  2.19041E+00  9.38158E-02  24.052

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)									
	8	CR53	2434	12	J2CR53	2.70133E-04	CR 2.19041E+00	1.08415E-02	24.053
	8	CR54	2437	13	J2CR54	6.72489E-05	CR 2.19041E+00	2.74986E-03	24.054
	8	NI58	2825	56	J2NI58	7.62173E-04	NI 2.19041E+00	3.34750E-02	28.058
	8	NI60	2831	57	J2NI60	2.93586E-04	NI 2.19041E+00	1.33386E-02	28.060
	8	NI61	2834	58	J2NI61	1.27631E-05	NI 2.19041E+00	5.89549E-04	28.061
	8	NI62	2837	59	J2NI62	4.06853E-05	NI 2.19041E+00	1.91008E-03	28.062
	8	NI64	2843	60	J2NI64	1.03673E-05	NI 2.19041E+00	5.02433E-04	28.064
	8	H1	10293	64	J2HINH20	5.52715E-02	H 2.19041E+00	4.22289E-02	1.001
	8	O16	825	33	J2O16	2.76358E-02	O 2.19041E+00	3.35102E-01	8.016
MATERIAL	9	FE54	2625	15	J2FE54	3.73343E-03	FE 7.92967E+00	4.21706E-02	26.054
	9	FE56	2631	66	J2FE56	5.80390E-02	FE 7.92967E+00	6.79825E-01	26.056
	9	FE57	2634	16	J2FE57	1.32885E-03	FE 7.92967E+00	1.58435E-02	26.057
	9	FE58	2637	17	J2FE58	1.77180E-04	FE 7.92967E+00	2.14949E-03	26.058
	9	CR50	2425	11	J2CR50	7.18316E-04	CR 7.92967E+00	7.51296E-03	24.050
	9	CR52	2431	53	J2CR52	1.38522E-02	CR 7.92967E+00	1.50667E-01	24.052
	9	CR53	2434	12	J2CR53	1.57054E-03	CR 7.92967E+00	1.74114E-02	24.053
	9	CR54	2437	13	J2CR54	3.90982E-04	CR 7.92967E+00	4.41624E-03	24.054
	9	NI58	2825	56	J2NI58	4.43124E-03	NI 7.92967E+00	5.37605E-02	28.058
	9	NI60	2831	57	J2NI60	1.70690E-03	NI 7.92967E+00	2.14216E-02	28.060
	9	NI61	2834	58	J2NI61	7.42043E-05	NI 7.92967E+00	9.46809E-04	28.061
	9	NI62	2837	59	J2NI62	2.36543E-04	NI 7.92967E+00	3.06756E-03	28.062
	9	NI64	2843	60	J2NI64	6.02748E-05	NI 7.92967E+00	8.06902E-04	28.064
MATERIAL	10	AL27	1325	1	J2AL27	3.35908E-02	AL 2.58326E+00	5.82597E-01	13.027
	10	B10	525	5	J2B10	7.09867E-03	B10 2.58326E+00	4.56899E-02	5.010
	10	B11	528	48	J2B11	3.92395E-02	B11 2.58326E+00	2.77693E-01	5.011
	10	C	600	6	J2C	1.21775E-02	C 2.58326E+00	9.40197E-02	6.000
MATERIAL	11	AL27	1325	1	J2AL27	6.02626E-02	AL 2.70000E+00	1.00000E+00	13.027
MATERIAL	12	FE54	2625	15	J2FE54	3.73343E-03	FE 7.92967E+00	4.21706E-02	26.054
	12	FE56	2631	66	J2FE56	5.80390E-02	FE 7.92967E+00	6.79825E-01	26.056
	12	FE57	2634	16	J2FE57	1.32885E-03	FE 7.92967E+00	1.58435E-02	26.057
	12	FE58	2637	17	J2FE58	1.77180E-04	FE 7.92967E+00	2.14949E-03	26.058
	12	CR50	2425	11	J2CR50	7.18316E-04	CR 7.92967E+00	7.51296E-03	24.050
	12	CR52	2431	53	J2CR52	1.38522E-02	CR 7.92967E+00	1.50667E-01	24.052
	12	CR53	2434	12	J2CR53	1.57054E-03	CR 7.92967E+00	1.74114E-02	24.053
	12	CR54	2437	13	J2CR54	3.90982E-04	CR 7.92967E+00	4.41624E-03	24.054
	12	NI58	2825	56	J2NI58	4.43124E-03	NI 7.92967E+00	5.37605E-02	28.058
	12	NI60	2831	57	J2NI60	1.70690E-03	NI 7.92967E+00	2.14216E-02	28.060
	12	NI61	2834	58	J2NI61	7.42043E-05	NI 7.92967E+00	9.46809E-04	28.061
	12	NI62	2837	59	J2NI62	2.36543E-04	NI 7.92967E+00	3.06756E-03	28.062
	12	NI64	2843	60	J2NI64	6.02748E-05	NI 7.92967E+00	8.06902E-04	28.064
MATERIAL	13	FE54	2625	15	J2FE54	3.73343E-03	FE 7.92967E+00	4.21706E-02	26.054
	13	FE56	2631	66	J2FE56	5.80390E-02	FE 7.92967E+00	6.79825E-01	26.056
	13	FE57	2634	16	J2FE57	1.32885E-03	FE 7.92967E+00	1.58435E-02	26.057
	13	FE58	2637	17	J2FE58	1.77180E-04	FE 7.92967E+00	2.14949E-03	26.058
	13	CR50	2425	11	J2CR50	7.18316E-04	CR 7.92967E+00	7.51296E-03	24.050
	13	CR52	2431	53	J2CR52	1.38522E-02	CR 7.92967E+00	1.50667E-01	24.052
	13	CR53	2434	12	J2CR53	1.57054E-03	CR 7.92967E+00	1.74114E-02	24.053
	13	CR54	2437	13	J2CR54	3.90982E-04	CR 7.92967E+00	4.41624E-03	24.054
	13	NI58	2825	56	J2NI58	4.43124E-03	NI 7.92967E+00	5.37605E-02	28.058
	13	NI60	2831	57	J2NI60	1.70690E-03	NI 7.92967E+00	2.14216E-02	28.060
	13	NI61	2834	58	J2NI61	7.42043E-05	NI 7.92967E+00	9.46809E-04	28.061
	13	NI62	2837	59	J2NI62	2.36543E-04	NI 7.92967E+00	3.06756E-03	28.062
	13	NI64	2843	60	J2NI64	6.02748E-05	NI 7.92967E+00	8.06902E-04	28.064
MATERIAL	14	AL27	1325	1	J2AL27	6.02626E-02	AL 2.70000E+00	1.00000E+00	13.027
MATERIAL	15	FE54	2625	15	J2FE54	3.73343E-03	FE 7.92967E+00	4.21706E-02	26.054
	15	FE56	2631	66	J2FE56	5.80390E-02	FE 7.92967E+00	6.79825E-01	26.056
	15	FE57	2634	16	J2FE57	1.32885E-03	FE 7.92967E+00	1.58435E-02	26.057
	15	FE58	2637	17	J2FE58	1.77180E-04	FE 7.92967E+00	2.14949E-03	26.058
	15	CR50	2425	11	J2CR50	7.18316E-04	CR 7.92967E+00	7.51296E-03	24.050
	15	CR52	2431	53	J2CR52	1.38522E-02	CR 7.92967E+00	1.50667E-01	24.052
	15	CR53	2434	12	J2CR53	1.57054E-03	CR 7.92967E+00	1.74114E-02	24.053
	15	CR54	2437	13	J2CR54	3.90982E-04	CR 7.92967E+00	4.41624E-03	24.054
	15	NI58	2825	56	J2NI58	4.43124E-03	NI 7.92967E+00	5.37605E-02	28.058
	15	NI60	2831	57	J2NI60	1.70690E-03	NI 7.92967E+00	2.14216E-02	28.060
	15	NI61	2834	58	J2NI61	7.42043E-05	NI 7.92967E+00	9.46809E-04	28.061
	15	NI62	2837	59	J2NI62	2.36543E-04	NI 7.92967E+00	3.06756E-03	28.062
	15	NI64	2843	60	J2NI64	6.02748E-05	NI 7.92967E+00	8.06902E-04	28.064
MATERIAL	16	FE54	2625	15	J2FE54	3.73343E-03	FE 7.92967E+00	4.21706E-02	26.054
	16	FE56	2631	66	J2FE56	5.80390E-02	FE 7.92967E+00	6.79825E-01	26.056
	16	FE57	2634	16	J2FE57	1.32885E-03	FE 7.92967E+00	1.58435E-02	26.057
	16	FE58	2637	17	J2FE58	1.77180E-04	FE 7.92967E+00	2.14949E-03	26.058
	16	CR50	2425	11	J2CR50	7.18316E-04	CR 7.92967E+00	7.51296E-03	24.050
	16	CR52	2431	53	J2CR52	1.38522E-02	CR 7.92967E+00	1.50667E-01	24.052
	16	CR53	2434	12	J2CR53	1.57054E-03	CR 7.92967E+00	1.74114E-02	24.053

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

	16	CR54	2437	13	J2CR54	3.90982E-04	CR	7.92967E+00	4.41624E-03	24.054
	16	NI58	2825	56	J2NI58	4.43124E-03	NI	7.92967E+00	5.37605E-02	28.058
	16	NI60	2831	57	J2NI60	1.70690E-03	NI	7.92967E+00	2.14216E-02	28.060
	16	NI61	2834	58	J2NI61	7.42043E-05	NI	7.92967E+00	9.46809E-04	28.061
	16	NI62	2837	59	J2NI62	2.36543E-04	NI	7.92967E+00	3.06756E-03	28.062
	16	NI64	2843	60	J2NI64	6.02748E-05	NI	7.92967E+00	8.06902E-04	28.064
MATERIAL	17	PB	8200	34	J2PB	3.20871E-02	PB	1.10400E+01	1.00000E+00	82.000
MATERIAL	18	B10	525	5	J2B10	8.55300E-05	B10	1.63187E+00	8.71455E-04	5.010
	18	B11	528	48	J2B11	3.42200E-04	B11	1.63187E+00	3.83359E-03	5.011
	18	AL27	1325	1	J2AL27	7.79600E-03	AL	1.63187E+00	2.14044E-01	13.027
	18	H1	10293	64	J2H1NH2O	5.85400E-02	H	1.63187E+00	6.00348E-02	1.001
	18	O16	825	33	J2O16	2.60900E-02	O	1.63187E+00	4.24640E-01	8.016
	18	C	600	6	J2C	2.26400E-02	C	1.63187E+00	2.76707E-01	6.000
	18	N14	725	31	J2N14	1.38884E-03	N	1.63187E+00	1.97898E-02	7.014
	18	N15	728	32	J2N15	5.15780E-06	N	1.63187E+00	7.87271E-05	7.015
MATERIAL	19	H1	10293	64	J2H1NH2O	6.67530E-02	H	9.98200E-01	1.11915E-01	1.001
	19	O16	825	33	J2O16	3.33765E-02	O	9.98200E-01	8.88085E-01	8.016
MATERIAL	20	H1	10293	64	J2H1NH2O	6.67530E-02	H	9.98200E-01	1.11915E-01	1.001
	20	O16	825	33	J2O16	3.33765E-02	O	9.98200E-01	8.88085E-01	8.016

BEGIN MATERIAL GEOMETRY

1

* ENTERING UNIT 2 TO READ MATERIAL GEOMETRY DATA. *

Changing input to named channel: CGF after reading
129 lines from named channel: INPUT Total line count= 335

qq NOTES ON PRINTING OF INPUT GEOMETRY DATA

Body and zone numbers in each part are local to that part.
Abbreviations for body parameters:
SPHERE R = Radius
BOX XL = Length on X axis; YL = Length on Y axis ; ZL = Length on Z axis
(X)ROD R = Radius ; H = Height of(X)axis.
(X)CONE RB = Radius of base ; RT = Radius of top ; H = Height of(X)axis.
TORUS RB = Bigger radius ; RS = Smaller radius.
(X)REL R = Radius ; HH = Half hgt.(X)axis.
(XY)PRISM B = Length of(X)base; H = Height of(Y)axis ; L = Prism length (Z);
Base angles : P = Origin end ; Q = Other end of base.
(X)SEC R1 = Inner radius ; R2 = Outer radius ; H = Height of(X)axis;
Angles from (Y) axis: T1 = lower theta ; T2 = Upper theta.
(X)HEMI R = Radius
(X)HEX D = Half-width across the hexagonal X-section; Height of (X) axis
RPP XU = Upper X limit ; YU = Upper Y limit ; ZU = Upper Z limit
All [R] = Matrix of direction cosines for rotated bodies:
Vx = new X axis ; Vy = New Y axis ; Vz = New Z axis

PART 1 (NEST) Bodies in earlier parts= 0

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX	1	0.0000	0.0000	0.0000	XL	21.6408	YL	21.6408
					ZL	348.1324		

Zone	Name	Contents	Region	Volume	Description (local bodies)
1		Body Hole	1	1.630E+05	+1
2	Exterior	(Extra zone created by code)			-1

PART 2 Was created by copying PART 1

	From	To	Count
Bodies	2 to	2 copied from bodies	1 to 1
Zones	3 to	4 copied from zones	1 to 2
Regions	2 to	2 replace regions	1 to 1

PART 3 (NEST) Bodies in earlier parts= 2

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX 1	0.0457	0.0000	0.3556	XL	0.1905	YL	20.2540 ZL 326.1360
The input characters				: [@b1t+@cvst]			
have been replaced by				: +.2362000			
BOX 2	0.0457	0.0000	0.0000	XL	0.1905	YL	20.2540 ZL 328.4220
BOX 3	0.0000	0.0000	0.0000	XL	0.2362	YL	20.2540 ZL 328.4220

Zone	Name	Contents	Region	Volume	Description (local bodies)
5		Body Hole	6	3	1.258E+03 +1
6		Material	3	4	8.820E+00 +2 -1
7		Material	9	5	3.040E+02 +3 -2
8	Exterior	(Extra zone created by code)			-3

PART 4 (NEST) Bodies in earlier parts= 5

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)			
The input characters				: [@wbr1+@overs]			
have been replaced by				: +21.27000			
The input characters				: [@wbr1+@overs]			
have been replaced by				: +21.27000			
BOX 1	0.0457	0.0000	0.3556	XL	0.1905	YL	21.2700 ZL 326.1360
The input characters				: [@b1t+@cvst]			
have been replaced by				: +.2362000			
The input characters				: [@wbr1+@overs]			
have been replaced by				: +21.27000			
BOX 2	0.0457	0.0000	0.0000	XL	0.1905	YL	21.2700 ZL 328.4220
BOX 3	0.0000	0.0000	0.0000	XL	0.2362	YL	21.2700 ZL 328.4220

Zone	Name	Contents	Region	Volume	Description (local bodies)
9		Body Hole	6	6	1.321E+03 +1
10		Material	3	7	9.263E+00 +2 -1
11		Material	9	8	3.192E+02 +3 -2
12	Exterior	(Extra zone created by code)			-3

The input characters		: [@ftth+@asoffx]
have been replaced by		: +.7061000
The input characters		: [@ftth+@asoffy]
have been replaced by		: +.7061000

PART 5 (FG PART) Bodies in earlier parts= 8

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX 1	0.7061	0.7061	0.0000	XL	21.6408	YL	21.6408 ZL 348.1324
The input characters				: [@baskoff+@lbw+@ftoffz]			
have been replaced by				: +5.080000			
BOX 2	0.1219	0.1219	0.0000	XL	22.2250	YL	22.2250 ZL 360.6800
The input characters				: [@wft+2*@ftth]			
have been replaced by				: +22.46880			
The input characters				: [@wft+2*@ftth]			
have been replaced by				: +22.46880			
BOX 3	0.0000	0.0000	5.0800	XL	22.4688	YL	22.4688 ZL 335.7880
The input characters				: [@wft+2*@ftth]			
have been replaced by				: +22.46880			
The input characters				: [@wft+2*@ftth]			
have been replaced by				: +22.46880			
BOX 4	0.0000	0.0000	0.0000	XL	22.4688	YL	22.4688 ZL 360.6800

Zone	Name	Contents	Region	Code	Vol.	Description (local bodies)
13	FUEL ASSEM	Part No.	1	-	1.630E+05	+1
14	SPACE IN T	Material	3	-	1.512E+04	+2 -1
15	FUEL TUBE	Material	9	-	-8.637E+03	+3 -2
16	CONTAINER	Material	3	-	-1.656E+05	+4 -3 -2
17	Exterior	(Extra zone created by code)				-4

There is a one-to-one correspondence between regions and zones.
The code volumes (calculated from zone descriptions and body volumes) will be used.
Zones 13 to 16 map to regions 9 to 12

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

The input characters : [ftth+asoffx]
have been replaced by : +.7061000
The input characters : [ftth+asoffyp]
have been replaced by : +.4140000

PART 6 (FG PART) Bodies in earlier parts= 12

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)		
BOX	1	0.7061	0.4140	0.0000	XL	21.6408	YL 21.6408 ZL 348.1324
The input characters : [baskoff+lbw+ftoffz]							
have been replaced by : +5.080000							
BOX	2	0.1219	0.1219	0.0000	XL	22.2250	YL 22.2250 ZL 360.6800
The input characters : [wft+2*ftth]							
have been replaced by : +22.46880							
The input characters : [wft+2*ftth]							
have been replaced by : +22.46880							
BOX	3	0.0000	0.0000	5.0800	XL	22.4688	YL 22.4688 ZL 335.7880
The input characters : [wft+2*ftth]							
have been replaced by : +22.46880							
The input characters : [wft+2*ftth]							
have been replaced by : +22.46880							
BOX	4	0.0000	0.0000	0.0000	XL	22.4688	YL 22.4688 ZL 360.6800

Zone	Name	Contents	Region	Code	Vol.	Description (local bodies)
18	FUEL ASSEM	Part No.	1	-	1.630E+05	+1
19	SPACE IN T	Material	3	-	1.512E+04	+2 -1
20	FUEL TUBE	Material	9	-	-8.637E+03	+3 -2
21	CONTAINER	Material	3	-	-1.656E+05	+4 -3 -2
22	Exterior	(Extra zone created by code)				-4

There is a one-to-one correspondence between regions and zones.
The code volumes (calculated from zone descriptions and body volumes) will be used.
Zones 18 to 21 map to regions 13 to 16
The input characters : [ftth+asoffxo]
have been replaced by : +1.722100
The input characters : [ftth+asoffyo]
have been replaced by : +1.722100

PART 7 (FG PART) Bodies in earlier parts= 16

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)		
BOX	1	1.7221	1.7221	0.0000	XL	21.6408	YL 21.6408 ZL 348.1324
The input characters : [wft+overs]							
have been replaced by : +23.24100							
The input characters : [wft+overs]							
have been replaced by : +23.24100							
The input characters : [baskoff+lbw+ftoffz]							
have been replaced by : +5.080000							
BOX	2	0.1219	0.1219	0.0000	XL	23.2410	YL 23.2410 ZL 360.6800
The input characters : [wft+2*ftth+overs]							
have been replaced by : +23.48480							
The input characters : [wft+2*ftth+overs]							
have been replaced by : +23.48480							
BOX	3	0.0000	0.0000	5.0800	XL	23.4848	YL 23.4848 ZL 335.7880
The input characters : [wft+2*ftth+overs]							
have been replaced by : +23.48480							
The input characters : [wft+2*ftth+overs]							
have been replaced by : +23.48480							
BOX	4	0.0000	0.0000	0.0000	XL	23.4848	YL 23.4848 ZL 360.6800

Zone	Name	Contents	Region	Code	Vol.	Description (local bodies)
23	FUEL ASSEM	Part No.	2	-	1.630E+05	+1
24	SPACE IN T	Material	3	-	3.178E+04	+2 -1
25	FUEL TUBE	Material	9	-	-9.620E+03	+3 -2
26	CONTAINER	Material	3	-	-1.811E+05	+4 -3 -2
27	Exterior	(Extra zone created by code)				-4

There is a one-to-one correspondence between regions and zones.
The code volumes (calculated from zone descriptions and body volumes) will be used.

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

Zones 23 to 26 map to regions 17 to 20

PART 8 (CLUSTER) Bodies in earlier parts= 20

Body definitions. Numbers are local to this part.

```

shape number  X0      Y0      Z0      Shape dependent parameters (see notes)
The input characters : [@tboffx+@cvst+@blt]
have been replaced by : +.6248000
The input characters : [@tboffy+@cvst+@blt]
have been replaced by : +.6248000
The input characters : [@wft+2*@ftth]
have been replaced by : +22.46880
The input characters : [@wft+2*@ftth]
have been replaced by : +22.46880
BOX 1 0.6248 0.6248 0.0000 XL 22.4688 YL 22.4688 ZL 360.6800
The input characters : [@tboffy+(@wfto+@wbrl)/2.0+@bshifty]
have been replaced by : +1.732250
The input characters : [@baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by : +7.112000
The input characters : [@blt+@cvst]
have been replaced by : +.2362000
The input characters : [@tboffx+@wfto]
have been replaced by : +23.32990
BOX 2 0.3886 1.7323 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
The input characters : [@tboffy+(@wfto+@wbrl)/2.0+@bshifty]
have been replaced by : +21.98625
The input characters : [@baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by : +7.112000
The input characters : [@blt+@cvst]
have been replaced by : +.2362000
BOX 3 23.3299 21.9862 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
[R]= Vx -1.00000 0.00000 0.00000
Vy 0.00000 -1.00000 0.00000
Vz 0.00000 0.00000 1.00000
The input characters : [@tboffx+(@wfto+@wbrl)/2.0+@bshifty]
have been replaced by : +1.732250
The input characters : [@tboffy+@wfto]
have been replaced by : +23.32990
The input characters : [@baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by : +7.112000
The input characters : [@blt+@cvst]
have been replaced by : +.2362000
BOX 4 1.7323 23.3299 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
[R]= Vx 0.00000 -1.00000 0.00000
Vy 1.00000 0.00000 0.00000
Vz 0.00000 0.00000 1.00000
The input characters : [@tboffx+(@wfto+@wbrl)/2.0+@bshifty]
have been replaced by : +21.98625
The input characters : [@baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by : +7.112000
The input characters : [@blt+@cvst]
have been replaced by : +.2362000
BOX 5 21.9862 0.3886 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
[R]= Vx 0.00000 1.00000 0.00000
Vy -1.00000 0.00000 0.00000
Vz 0.00000 0.00000 1.00000
BOX 6 0.0000 0.0000 0.0000 XL 23.3299 YL 23.3299 ZL 360.6800

```

Zone	Name	Contents	Region	Volume	Description (local bodies)
28	Part No.	5	21	1.821E+05	+1
29	Part No.	3	22	1.571E+03	+2
30	Part No.	3	23	1.571E+03	+3
31	Part No.	3	24	1.571E+03	+4
32	Part No.	3	25	1.571E+03	+5
33	Material	3	26	7.940E+03	+6 -1 -2 -3 -4 -5
34	Exterior	(Extra zone created by code)			-6

PART 9 (CLUSTER) Bodies in earlier parts= 26

Body definitions. Numbers are local to this part.

```

shape number  X0      Y0      Z0      Shape dependent parameters (see notes)

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```

The input characters      : [tboffx+@cvst+@blt]
have been replaced by    : +.6248000
The input characters      : [tboffyp+@cvst+@blt]
have been replaced by    : +.4305000
The input characters      : [wft+2*@ftth]
have been replaced by    : +22.46880
The input characters      : [wft+2*@ftth]
have been replaced by    : +22.46880
BOX 1 0.6248 0.4305 0.0000 XL 22.4688 YL 22.4688 ZL 360.6800
The input characters      : [tboffyp+(@wfto-@wbrl)/2.0+@bshiftp]
have been replaced by    : +1.537950
The input characters      : [baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by    : +7.112000
The input characters      : [blt+@cvst]
have been replaced by    : +.2362000
The input characters      : [tboffx+@wfto]
have been replaced by    : +23.32990
BOX 2 0.3886 1.5379 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
The input characters      : [tboffyp+(@wfto+@wbrl)/2.0+@bshiftp]
have been replaced by    : +21.79195
The input characters      : [baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by    : +7.112000
The input characters      : [blt+@cvst]
have been replaced by    : +.2362000
BOX 3 23.3299 21.7919 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
Vx -1.00000 0.00000 0.00000
[R]= Vy 0.00000 -1.00000 0.00000
Vz 0.00000 0.00000 1.00000
The input characters      : [tboffx+(@wfto-@wbrl)/2.0+@bshiftp]
have been replaced by    : +1.732250
The input characters      : [tboffyp+@wfto]
have been replaced by    : +23.13560
The input characters      : [baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by    : +7.112000
The input characters      : [blt+@cvst]
have been replaced by    : +.2362000
BOX 4 1.7323 23.1356 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
Vx 0.00000 -1.00000 0.00000
[R]= Vy 1.00000 0.00000 0.00000
Vz 0.00000 0.00000 1.00000
The input characters      : [tboffx+(@wfto+@wbrl)/2.0+@bshiftp]
have been replaced by    : +21.98625
The input characters      : [baskoff+@lbw+@ftoffz+@cvoffz]
have been replaced by    : +7.112000
The input characters      : [blt+@cvst]
have been replaced by    : +.2362000
BOX 5 21.9862 0.1943 7.1120 XL 0.2362 YL 20.2540 ZL 328.4220
Vx 0.00000 1.00000 0.00000
[R]= Vy -1.00000 0.00000 0.00000
Vz 0.00000 0.00000 1.00000
BOX 6 0.0000 0.0000 0.0000 XL 23.3299 YL 23.3299 ZL 360.6800

Zone Name Contents Region Volume Description (local bodies)
35 Part No. 6 27 1.821E+05 +1
36 Part No. 3 28 1.571E+03 +2
37 Part No. 3 29 1.571E+03 +3
38 Part No. 3 30 1.571E+03 +4
39 Part No. 3 31 1.571E+03 +5
40 Material 3 32 7.940E+03 +6 -1 -2 -3 -4 -5
41 Exterior (Extra zone created by code) -6

PART 10 (CLUSTER) Bodies in earlier parts= 32
-----
Body definitions. Numbers are local to this part.

shape number X0 Y0 Z0 Shape dependent parameters (see notes)
The input characters      : [tboffx+@cvst+@blt]
have been replaced by    : +.6248000
The input characters      : [tboffyp+@cvst+@blt]
have been replaced by    : +.6248000
The input characters      : [wft+2*@ftth+@overs]
have been replaced by    : +23.48480
The input characters      : [wft+2*@ftth+@overs]
have been replaced by    : +23.48480

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

```

BOX      1      0.6248      0.6248      0.0000 XL      23.4848 YL      23.4848 ZL      360.6800
The input characters      :  { @tboffy+(@wfto-@wbrl)/2.0+@bshifty}
have been replaced by    :  +1.732250
The input characters      :  { @baskoff+@lbw+@ftoffz+@cvoffz}
have been replaced by    :  +7.112000
The input characters      :  { @blt+@cvst}
have been replaced by    :  +.2362000
The input characters      :  { @wbrl+@overs}
have been replaced by    :  +21.27000
The input characters      :  { @tboffx+@wfto+@overs}
have been replaced by    :  +24.34590
BOX      2      0.3886      1.7323      7.1120 XL      0.2362 YL      21.2700 ZL      328.4220
The input characters      :  { @tboffy+(@wfto-@wbrl)/2.0+@overs+@bshifty}
have been replaced by    :  +23.00225
The input characters      :  { @baskoff+@lbw+@ftoffz+@cvoffz}
have been replaced by    :  +7.112000
The input characters      :  { @blt+@cvst}
have been replaced by    :  +.2362000
The input characters      :  { @wbrl+@overs}
have been replaced by    :  +21.27000
BOX      3      24.3459      23.0023      7.1120 XL      0.2362 YL      21.2700 ZL      328.4220
[R]=      Vx      -1.00000      0.00000      0.00000
          Vy      0.00000      -1.00000      0.00000
          Vz      0.00000      0.00000      1.00000
The input characters      :  { @tboffx+(@wfto-@wbrl)/2.0+@bshiftyx}
have been replaced by    :  +1.732250
The input characters      :  { @tboffy+@wfto+@overs}
have been replaced by    :  +24.34590
The input characters      :  { @baskoff+@lbw+@ftoffz+@cvoffz}
have been replaced by    :  +7.112000
The input characters      :  { @blt+@cvst}
have been replaced by    :  +.2362000
The input characters      :  { @wbrl+@overs}
have been replaced by    :  +21.27000
BOX      4      1.7323      24.3459      7.1120 XL      0.2362 YL      21.2700 ZL      328.4220
[R]=      Vx      0.00000      -1.00000      0.00000
          Vy      1.00000      0.00000      0.00000
          Vz      0.00000      0.00000      1.00000
The input characters      :  { @tboffx+(@wfto-@wbrl)/2.0+@overs+@bshiftyx}
have been replaced by    :  +23.00225
The input characters      :  { @baskoff+@lbw+@ftoffz+@cvoffz}
have been replaced by    :  +7.112000
The input characters      :  { @blt+@cvst}
have been replaced by    :  +.2362000
The input characters      :  { @wbrl+@overs}
have been replaced by    :  +21.27000
BOX      5      23.0023      0.3886      7.1120 XL      0.2362 YL      21.2700 ZL      328.4220
[R]=      Vx      0.00000      1.00000      0.00000
          Vy      -1.00000      0.00000      0.00000
          Vz      0.00000      0.00000      1.00000
The input characters      :  { @opspd+@overs}
have been replaced by    :  +24.34590
The input characters      :  { @opspd+@overs}
have been replaced by    :  +24.34590
BOX      6      0.0000      0.0000      0.0000 XL      24.3459 YL      24.3459 ZL      360.6800

Zone  Name      Contents      Region      Volume      Description (local bodies)
42      Part No.      7      33      1.989E+05      +1
43      Part No.      4      34      1.650E+03      +2
44      Part No.      4      35      1.650E+03      +3
45      Part No.      4      36      1.650E+03      +4
46      Part No.      4      37      1.650E+03      +5
47      Material      3      38      8.255E+03      +6      -1      -2      -3      -4      -5
48      Exterior      (Extra zone created by code)      -6

PART 11 (CLUSTER) Bodies in earlier parts= 38.
-----
Body definitions. Numbers are local to this part.

shape number  X0      Y0      Z0      Shape dependent parameters (see notes)
The input characters      :  [-1.0*(@wsspxo+(@opspd+@overs)/2.0)]
have been replaced by    :  -52.30005
The input characters      :  { @tespyo-(@opspd+@overs)/2.0}
have been replaced by    :  +41.88605

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@opspd+@overs]			
have been replaced by	:	+24.34590			
The input characters	:	[@opspd+@overs]			
have been replaced by	:	+24.34590			
The input characters	:	[-1.0*(@wsspx1+@opspd/2.0)]			
have been replaced by	:	-25.19825			
The input characters	:	[@tsspy2-@opspd/2.0]			
have been replaced by	:	+47.60105			
BOX 1 -52.3000		41.8860 -15.4686 XL 24.3459 YL 24.3459 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@wsspx1-@opspd/2.0]			
have been replaced by	:	+1.868350			
The input characters	:	[@tsspy2-@opspd/2.0]			
have been replaced by	:	+47.60105			
BOX 2 -25.1983		47.6011 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@wsspxo-(@opspd+@overs)/2.0]			
have been replaced by	:	+27.95415			
BOX 3 1.8683		47.6011 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[@tsspyo-(@opspd+@overs)/2.0]			
have been replaced by	:	+41.88605			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@opspd+@overs]			
have been replaced by	:	+24.34590			
The input characters	:	[@opspd+@overs]			
have been replaced by	:	+24.34590			
The input characters	:	[-1.0*(@wsspx3+@opspd/2.0)]			
have been replaced by	:	-77.11584			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 4 27.9541		41.8860 -15.4686 XL 24.3459 YL 24.3459 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[-1.0*(@wsspx2+@opspd/2.0)]			
have been replaced by	:	-51.28405			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 5 -77.1158		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[-1.0*(@wsspx1+@opspd/2.0)]			
have been replaced by	:	-25.19825			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 6 -51.2841		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@wsspx1-@opspd/2.0]			
have been replaced by	:	+1.868350			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 7 -25.1983		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@wsspx2-@opspd/2.0]			
have been replaced by	:	+27.95415			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 8 1.8683		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[@wsspx3-@opspd/2.0]			
have been replaced by	:	+53.78595			
The input characters	:	[@tsspy1-@opspd/2.0]			
have been replaced by	:	+15.41925			
BOX 9 27.9541		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800			
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]			
have been replaced by	:	-15.46860			
The input characters	:	[-1.0*(@wsspx3+@opspd/2.0)]			

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

have been replaced by	:	-77.11584				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 10 53.7859		15.4192 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[-1.0*(@wsspx2+@opspd/2.0)]				
have been replaced by	:	-51.28405				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 11 -77.1158		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[-1.0*(@wsspx1+@opspd/2.0)]				
have been replaced by	:	-25.19825				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 12 -51.2841		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[@wsspx1-@opspd/2.0]				
have been replaced by	:	+1.868350				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 13 -25.1983		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[@wsspx2-@opspd/2.0]				
have been replaced by	:	+27.95415				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 14 1.8683		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[@wsspx3-@opspd/2.0]				
have been replaced by	:	+53.78595				
The input characters	:	[-1.0*(@opspd/2.0)]				
have been replaced by	:	-11.66495				
BOX 15 27.9541		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[-1.0*(@wsspx3+@opspd/2.0)]				
have been replaced by	:	-77.11584				
The input characters	:	[-1.0*(@tsspy1+@opspd/2.0)]				
have been replaced by	:	-38.74915				
BOX 16 53.7859		-11.6649 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[-1.0*(@wsspx2+@opspd/2.0)]				
have been replaced by	:	-51.28405				
The input characters	:	[-1.0*(@tsspy1+@opspd/2.0)]				
have been replaced by	:	-38.74915				
BOX 17 -77.1158		-38.7492 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[-1.0*(@wsspx1+@opspd/2.0)]				
have been replaced by	:	-25.19825				
The input characters	:	[-1.0*(@tsspy1+@opspd/2.0)]				
have been replaced by	:	-38.74915				
BOX 18 -51.2841		-38.7492 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[@wsspx1-@opspd/2.0]				
have been replaced by	:	+1.868350				
The input characters	:	[-1.0*(@tsspy1+@opspd/2.0)]				
have been replaced by	:	-38.74915				
BOX 19 -25.1983		-38.7492 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				
The input characters	:	[-1.0*(@baskoff+@lbw+@lbws)]				
have been replaced by	:	-15.46860				
The input characters	:	[@wsspx2-@opspd/2.0]				
have been replaced by	:	+27.95415				
The input characters	:	[-1.0*(@tsspy1+@opspd/2.0)]				
have been replaced by	:	-38.74915				
BOX 20 1.8683		-38.7492 -15.4686 XL 23.3299 YL 23.3299 ZL 360.6800				

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

```

The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [@wsspx3-@opspd/2.0]
have been replaced by    : +53.78595
The input characters      : [-1.0*(@tsspy1+@opspd/2.0)]
have been replaced by    : -38.74915
BOX      21      27.9541  -38.7492  -15.4686 XL   23.3299 YL   23.3299 ZL   360.6800
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [-1.0*(@wsspxo+(@opspd+@overs)/2.0)]
have been replaced by    : -52.30005
BOX      22      53.7859  -38.7492  -15.4686 XL   23.3299 YL   23.3299 ZL   360.6800
The input characters      : [-1.0*(@tsspyo+(@opspd+@overs)/2.0)]
have been replaced by    : -66.23195
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [@opspd+@overs]
have been replaced by    : +24.34590
The input characters      : [@opspd+@overs]
have been replaced by    : +24.34590
The input characters      : [-1.0*(@wsspx1+@opspd/2.0)]
have been replaced by    : -25.19825
The input characters      : [-1.0*(@tsspy2+@opspd/2.0)]
have been replaced by    : -70.93095
BOX      23      -52.3000  -66.2319  -15.4686 XL   24.3459 YL   24.3459 ZL   360.6800
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [@wsspx1-@opspd/2.0]
have been replaced by    : +1.868350
The input characters      : [-1.0*(@tsspy2+@opspd/2.0)]
have been replaced by    : -70.93095
BOX      24      -25.1983  -70.9309  -15.4686 XL   23.3299 YL   23.3299 ZL   360.6800
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [@wsspxo-(@opspd+@overs)/2.0]
have been replaced by    : +27.95415
BOX      25      1.8683    -70.9309  -15.4686 XL   23.3299 YL   23.3299 ZL   360.6800
The input characters      : [-1.0*(@tsspyo+(@opspd+@overs)/2.0)]
have been replaced by    : -66.23195
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
The input characters      : [@opspd+@overs]
have been replaced by    : +24.34590
The input characters      : [@opspd+@overs]
have been replaced by    : +24.34590
The input characters      : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by    : -15.46860
BOX      26      27.9541  -66.2319  -15.4686 XL   24.3459 YL   24.3459 ZL   360.6800
The input characters      : [@diaht/2.0]
have been replaced by    : +87.46490
ZROD     27      0.0000    0.0000  -15.4686 R    87.4649 H    360.6800

```

Zone	Name	Contents	Region	Volume	Description (local bodies)
49	Part No.	16	39	2.138E+05	+1
50	Part No.	12	40	1.963E+05	+2
51	Part No.	13	41	1.963E+05	+3
52	Part No.	17	42	2.138E+05	+4
53	Part No.	12	43	1.963E+05	+5
54	Part No.	12	44	1.963E+05	+6
55	Part No.	12	45	1.963E+05	+7
56	Part No.	13	46	1.963E+05	+8
57	Part No.	13	47	1.963E+05	+9
58	Part No.	13	48	1.963E+05	+10
59	Part No.	9	49	1.963E+05	+11
60	Material	3	50	1.963E+05	+12
61	Part No.	9	51	1.963E+05	+13
62	Part No.	15	52	1.963E+05	+14
63	Material	3	53	1.963E+05	+15
64	Part No.	15	54	1.963E+05	+16
65	Part No.	8	55	1.963E+05	+17
66	Part No.	8	56	1.963E+05	+18
67	Part No.	8	57	1.963E+05	+19
68	Part No.	14	58	1.963E+05	+20
69	Part No.	14	59	1.963E+05	+21

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

70	Part No.	14	60	1.963E+05	+22						
71	Part No.	10	61	2.138E+05	+23						
72	Part No.	8	62	1.963E+05	+24						
73	Part No.	14	63	1.963E+05	+25						
74	Part No.	18	64	2.138E+05	+26						
75	Hole Mat.	7	65	3.494E+06	+27	-1	-2	-3	-4	-5	-6
						-7	-8	-9	-10	-11	-12
						-14	-15	-16	-17	-18	-19
						-21	-22	-23	-24	-25	-26
76	Exterior	(Extra zone created by code)			-27						

PART 12 (NEST) Bodies in earlier parts= 65

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX	1	0.0000	23.3299	0.0000	XL	23.3299	YL	23.3299
		Vx	0.00000	-1.00000	0.00000			
[R]=		Vy	1.00000	0.00000	0.00000			
		Vz	0.00000	0.00000	1.00000			
BOX	2	0.0000	0.0000	0.0000	XL	23.3299	YL	23.3299
						ZL	360.6800	

Zone	Name	Contents	Region	Volume	Description (local bodies)
77		Part No.	8	66	1.963E+05
78		Material	3	67	0.000E+00
79	Exterior	(Extra zone created by code)			-2

PART 13 (NEST) Bodies in earlier parts= 67

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX	1	23.3299	23.3299	0.0000	XL	23.3299	YL	23.3299
		Vx	-1.00000	0.00000	0.00000			
[R]=		Vy	0.00000	-1.00000	0.00000			
		Vz	0.00000	0.00000	1.00000			
BOX	2	0.0000	0.0000	0.0000	XL	23.3299	YL	23.3299
						ZL	360.6800	

Zone	Name	Contents	Region	Volume	Description (local bodies)
80		Part No.	8	68	1.963E+05
81		Material	3	69	0.000E+00
82	Exterior	(Extra zone created by code)			-2

PART 14 (NEST) Bodies in earlier parts= 69

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX	1	23.3299	0.0000	0.0000	XL	23.3299	YL	23.3299
		Vx	0.00000	1.00000	0.00000			
[R]=		Vy	-1.00000	0.00000	0.00000			
		Vz	0.00000	0.00000	1.00000			
BOX	2	0.0000	0.0000	0.0000	XL	23.3299	YL	23.3299
						ZL	360.6800	

Zone	Name	Contents	Region	Volume	Description (local bodies)
83		Part No.	8	70	1.963E+05
84		Material	3	71	0.000E+00
85	Exterior	(Extra zone created by code)			-2

PART 15 (NEST) Bodies in earlier parts= 71

Body definitions. Numbers are local to this part.

shape	number	X0	Y0	Z0	Shape dependent parameters (see notes)			
BOX	1	23.3299	23.3299	0.0000	XL	23.3299	YL	23.3299
		Vx	-1.00000	0.00000	0.00000			
[R]=		Vy	0.00000	-1.00000	0.00000			
		Vz	0.00000	0.00000	1.00000			
BOX	2	0.0000	0.0000	0.0000	XL	23.3299	YL	23.3299
						ZL	360.6800	

Zone	Name	Contents	Region	Volume	Description (local bodies)
86		Part No.	9	72	1.963E+05
87		Material	3	73	0.000E+00
88	Exterior	(Extra zone created by code)			-2

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

PART 16 (NEST) Bodies in earlier parts= 73

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
BOX	1	0.0000	24.3459	0.0000 XL	24.3459 YL	24.3459 ZL	360.6800				
	Vx	0.00000	-1.00000	0.00000							
[R]=	Vy	1.00000	0.00000	0.00000							
	Vz	0.00000	0.00000	1.00000							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
BOX	2	0.0000	0.0000	0.0000 XL	24.3459 YL	24.3459 ZL	360.6800				

Zone	Name	Contents	Region	Volume	Description (local bodies)
89		Part No.	10	74	2.138E+05 +1
90		Material	3	75	0.000E+00 +2 -1
91	Exterior	(Extra zone created by code)			-2

PART 17 (NEST) Bodies in earlier parts= 75

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
BOX	1	24.3459	24.3459	0.0000 XL	24.3459 YL	24.3459 ZL	360.6800				
	Vx	-1.00000	0.00000	0.00000							
[R]=	Vy	0.00000	-1.00000	0.00000							
	Vz	0.00000	0.00000	1.00000							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
BOX	2	0.0000	0.0000	0.0000 XL	24.3459 YL	24.3459 ZL	360.6800				

Zone	Name	Contents	Region	Volume	Description (local bodies)
92		Part No.	10	76	2.138E+05 +1
93		Material	3	77	0.000E+00 +2 -1
94	Exterior	(Extra zone created by code)			-2

PART 18 (NEST) Bodies in earlier parts= 77

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										
BOX	1	24.3459	0.0000	0.0000 XL	24.3459 YL	24.3459 ZL	360.6800				
	Vx	0.00000	1.00000	0.00000							
[R]=	Vy	-1.00000	0.00000	0.00000							
	Vz	0.00000	0.00000	1.00000							
The input characters	: [@opspd+@overs]										
have been replaced by	: +24.34590										

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

The input characters : [@opspd+@overs]
have been replaced by : +24.34590
BOX 2 0.0000 0.0000 0.0000 XL 24.3459 YL 24.3459 ZL 360.6800

Zone	Name	Contents	Region	Volume	Description (local bodies)
95		Part No. 10	78	2.138E+05	+1
96		Material 3	79	0.000E+00	+2 -1
97	Exterior	(Extra zone created by code)			-2

PART 19 (NEST) Bodies in earlier parts= 79

Body definitions. Numbers are local to this part.

shape number X0 Y0 Z0 Shape dependent parameters (see notes)
The input characters : [@diaht/2.0]
have been replaced by : +87.46490
The input characters : [@canod/2.0-@canth]
have been replaced by : +88.12530
ZROD 1 0.0000 0.0000 0.0000 R 87.4649 H 360.6800
ZROD 2 0.0000 0.0000 0.0000 R 88.1253 H 360.6800

Zone	Name	Contents	Region	Volume	Description (local bodies)
98		Part No. 11	80	8.668E+06	+1
99		Material 3	81	1.314E+05	+2 -1
100	Exterior	(Extra zone created by code)			-2

The input characters : [@canod/2.0-@canth]
have been replaced by : +88.12530

PART 20 (FG PART) Bodies in earlier parts= 81

Body definitions. Numbers are local to this part.

shape number X0 Y0 Z0 Shape dependent parameters (see notes)
The input characters : [-1*@canbot]
have been replaced by : -4.445000
The input characters : [@canod/2.0]
have been replaced by : +89.71280
ZROD 1 0.0000 0.0000 0.0000 R 88.1253 H 360.6800
The input characters : [@canod/2.0-@canth]
have been replaced by : +88.12530
ZROD 2 0.0000 0.0000 -4.4450 R 89.7128 H 4.4450
The input characters : [@cavheight+@shield1]
have been replaced by : +3.7338000E+02
The input characters : [@canod/2.0-@canth]
have been replaced by : +88.12530
ZROD 3 0.0000 0.0000 360.6800 R 88.1253 H 12.7000
The input characters : [@canod/2.0-@canth]
have been replaced by : +88.12530
The input characters : [@can1-@canbot]
have been replaced by : +3.8100000E+02
ZROD 4 0.0000 0.0000 373.3800 R 88.1253 H 7.6200
The input characters : [@canod/2.0]
have been replaced by : +89.71280
The input characters : [@can1-@canbot]
have been replaced by : +3.8100000E+02
ZROD 5 0.0000 0.0000 0.0000 R 88.1253 H 381.0000
The input characters : [-1*@canbot]
have been replaced by : -4.445000
The input characters : [@canod/2.0]
have been replaced by : +89.71280
ZROD 6 0.0000 0.0000 0.0000 R 89.7128 H 381.0000
ZROD 7 0.0000 0.0000 -4.4450 R 89.7128 H 385.4450

Zone	Name	Contents	Region	Code Vol.	Description (local bodies)
101	CAVITY	Part No. 19	-	8.800E+06	+1
102	BOTTOMPLAT	Material 15	-	1.124E+05	+2
103	SHIELDLID	Material 15	-	3.099E+05	+3
104	STRUCTLID	Material 15	-	1.859E+05	+4
105	SHELL	Material 15	-	3.379E+05	+6 -5

Returning input to named channel: INPUT after reading
422 lines from named channel: CGF Total line count= 756

The input characters : [@tscNUM + 1]
have been replaced by : +21.00000

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

106 CANISTER Material 0 - -1.859E+05 +7 -6 -4 -2
107 Exterior (Extra zone created by code) -7

There is a one-to-one correspondence between regions and zones.
The code volumes (calculated from zone descriptions and body volumes) will be used.
Zones 101 to 106 map to regions 82 to 87
The input characters : [cpi*@tscOD/2]
have been replaced by : +89.71280
The input characters : [cpi*@tscHT]
have been replaced by : +3.854450E+02

PART 21 (FG PART) Bodies in earlier parts= 88

Body definitions. Numbers are local to this part.

shape number	X0	Y0	Z0	Shape dependent parameters (see notes)
The input characters	: [cpi*@inshlID/2]			
have been replaced by	: +90.80500			
The input characters	: [cpi*(@btmplHT+@inshlHT+@topplHT)]			
have been replaced by	: +3.8862000E+02			
ZROD 1	0.0000	0.0000	R	89.7128 H 385.4450
The input characters	: [cpi*@tfrOD/2]			
have been replaced by	: +1.1303000E+02			
The input characters	: [cpi*@btmplHT]			
have been replaced by	: +2.540000			
ZROD 2	0.0000	0.0000	R	90.8050 H 388.6200
The input characters	: [cpi*@btmplHT]			
have been replaced by	: +2.540000			
The input characters	: [cpi*(@inshlID/2)+@inshlTH]			
have been replaced by	: +92.71000			
ZROD 3	0.0000	0.0000	R	113.0300 H 2.5400
The input characters	: [cpi*@inshlHT]			
have been replaced by	: +3.8100000E+02			
The input characters	: [cpi*@btmplHT]			
have been replaced by	: +2.540000			
The input characters	: [cpi*(@inshlID/2)+@inshlTH+@pshlTH]			
have been replaced by	: +1.0287000E+02			
ZROD 4	0.0000	0.0000	R	92.7100 H 381.0000
The input characters	: [cpi*(@inshlHT-@ns4frGAP)]			
have been replaced by	: +3.7084000E+02			
The input characters	: [cpi*@btmplHT]			
have been replaced by	: +2.540000			
The input characters	: [cpi*(@tfrOD/2)-@otshlTH]			
have been replaced by	: +1.0985500E+02			
ZROD 5	0.0000	0.0000	R	102.8700 H 370.8400
The input characters	: [cpi*@inshlHT]			
have been replaced by	: +3.8100000E+02			
The input characters	: [cpi*@btmplHT]			
have been replaced by	: +2.540000			
The input characters	: [cpi*(@tfrOD/2)]			
have been replaced by	: +1.1303000E+02			
The input characters	: [cpi*@inshlHT]			
have been replaced by	: +3.8100000E+02			
ZROD 6	0.0000	0.0000	R	109.8550 H 381.0000
The input characters	: [cpi*(@btmplHT+@inshlHT)]			
have been replaced by	: +3.8354001E+02			
The input characters	: [cpi*(@tfrOD/2)]			
have been replaced by	: +1.1303000E+02			
ZROD 7	0.0000	0.0000	R	113.0300 H 381.0000
The input characters	: [cpi*@topplHT]			
have been replaced by	: +5.080000			
The input characters	: [cpi*(@btmplHT+@inshlHT+@topplHT)]			
have been replaced by	: +3.8862000E+02			
The input characters	: [cpi*@rtnrnID/2]			
have been replaced by	: +86.99500			
ZROD 8	0.0000	0.0000	R	113.0300 H 5.0800
The input characters	: [cpi*@rtnrnHT]			
have been replaced by	: +1.905000			
The input characters	: [cpi*(@btmplHT+@inshlHT+@topplHT)]			
have been replaced by	: +3.8862000E+02			
The input characters	: [cpi*@rtnrnOD/2]			
have been replaced by	: +1.0261600E+02			
ZROD 9	0.0000	0.0000	R	86.9950 H 1.9050
The input characters	: [cpi*@rtnrnHT]			

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```

have been replaced by      : +1.905000
The input characters       : [-shldrHT*@cpi]
have been replaced by      : -24.13000
The input characters       : [@cpi*@tfrOD/2]
have been replaced by      : +1.1303000E+02
The input characters       : [@cpi*@shldrHT]
have been replaced by      : +24.13000
ZROD 10 0.0000 0.0000 388.6200 R 102.6160 H 1.9050
The input characters       : [@cpi*((@drrailID/2)+@drrailWD)]
have been replaced by      : +1.0731499E+02
ZROD 11 0.0000 0.0000 -24.1300 R 113.0300 H 24.1300
The input characters       : [-@cpi*((@drrailID/2)+@drrailWD)]
have been replaced by      : -1.0731499E+02
YP 12 At 107.3150
The input characters       : [-@cpi*@tfrOD/2]
have been replaced by      : -1.1303000E+02
YP 13 At -107.3150
The input characters       : [@cpi*(@btmplHT+@inshlHT+@topplHT - @trunzcor)]
have been replaced by      : +3.5051999E+02
The input characters       : [@cpi*@trunOD/2]
have been replaced by      : +12.70000
The input characters       : [@cpi*@tfrOD]
have been replaced by      : +2.260600E+02
The input characters       : [-1 * @cpi*@shldrHT]
have been replaced by      : -24.13000
The input characters       : [@cpi*@tfrOD/2]
have been replaced by      : +1.1303000E+02
XROD 14 -113.0300 0.0000 350.5200 R 12.7000 H 226.0600
The input characters       : [@cpi*(@shldrHT+@btmplHT+@inshlHT+@topplHT+@rttrnHT)]
have been replaced by      : +4.1465500E+02
ZROD 15 0.0000 0.0000 -24.1300 R 113.0300 H 414.6550

Zone Name Contents Region Code Vol. Description (local bodies)
The input characters : [@tscMAT + @cavMAT]
have been replaced by : +19.00000
108 Part No. 20 - 9.746E+06 +1
The input characters : [@tscMAT + @btmplMAT]
have been replaced by : +16.00000
109 Material 19 - 3.210E+05 +2 -1
The input characters : [@tscMAT + @inshlMAT]
have been replaced by : +16.00000
110 Material 16 - -9.965E+06 +3 -2
The input characters : [@tscMAT + @pbshlMAT]
have been replaced by : +17.00000
111 Material 16 - 1.065E+05 +4 -2 -14
The input characters : [@tscMAT + @n4shlMAT]
have been replaced by : +18.00000
112 Material 17 - 1.926E+06 +5 -4 -14
The input characters : [@tscMAT + @otshlMAT]
have been replaced by : +16.00000
113 Material 18 - -8.286E+06 +6 -5 -4 -14
The input characters : [@tscMAT + @topplMAT]
have been replaced by : +16.00000
114 Material 16 - 7.325E+05 +7 -6 -14
The input characters : [@tscMAT + @cavMAT]
have been replaced by : +19.00000
115 Material 16 - -9.863E+06 +8 -2
The input characters : [@tscMAT + @rttrnMAT]
have been replaced by : +16.00000
116 Material 19 - 4.529E+04 +9
The input characters : [@tscMAT + @shldrMAT]
have been replaced by : +16.00000
117 Material 16 - 1.773E+04 +10 -9
The input characters : [@tscMAT + @cavMAT]
have been replaced by : +19.00000
118 Material 16 - 9.685E+05 +11 -12 +13
The input characters : [@tscMAT + @cavMAT]
have been replaced by : +19.00000
119 Material 19 - 9.685E+05 +11 +12
The input characters : [@tscMAT + @trunMAT]
have been replaced by : +16.00000
120 Material 19 - 9.685E+05 +11 -13
The input characters : [@tscMAT + @cavMAT]
have been replaced by : +19.00000

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

121	Material	16	-	6.690E+06	+14	+15	-2			
122	Material	19	-	-1.011E+05	+15	-14	-11	-10	-8	-7 -3
123	Exterior	(Extra zone created by code)			-15					

There is a one-to-one correspondence between regions and zones.
The code volumes (calculated from zone descriptions and body volumes) will be used.
Zones 108 to 122 map to regions 88 to 102

The input data has been expanded into a global model in which each occurrence of a part becomes unique. The parts and bodies in this expanded scheme are renumbered and referred to as 'Assembled' parts and bodies

Summary of part structure
The global part is input part number 21
Map of part structure. Numbers are input parts.

```

21 Contains:
  20 Contains:
    19 Contains:
      11 Contains:
        16 Contains:
          10 Contains:
            7 Contains:
              2 s.p.
              4 s.p.
              4 s.p.
              4 s.p.
              4 s.p.
            12 Contains:
              8 Contains:
                5 Contains:
                  1 s.p.
                  3 s.p.
                  3 s.p.
                  3 s.p.
                  3 s.p.
              13 Contains:
                8 etc.
                5 etc.
              17 Contains:
                10 etc.
                7 etc.
              12 etc.
                8 etc.
                5 etc.
              12 etc.
                8 etc.
                5 etc.
              12 etc.
                8 etc.
                5 etc.
              13 etc.
                8 etc.
                5 etc.
              13 etc.
                8 etc.
                5 etc.
              13 etc.
                8 etc.
                5 etc.
              9 Contains:
                6 Contains:
                  1 s.p.
                  3 s.p.
                  3 s.p.
                  3 s.p.
                  3 s.p.
              9 etc.
                6 etc.
              15 Contains:
                9 etc.
                6 etc.
              15 etc.
                9 etc.

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```

      6 etc.
      8 etc.
      5 etc.
      8 etc.
      5 etc.
      8 etc.
      5 etc.
      14 Contains:
      8 etc.
      5 etc.
      14 etc.
      8 etc.
      5 etc.
      14 etc.
      8 etc.
      5 etc.
      10 etc.
      7 etc.
      8 etc.
      5 etc.
      14 etc.
      8 etc.
      5 etc.
      18 Contains:
      10 etc.
      7 etc.

Maximum level of nesting           = 8
Number of assembled parts          = 189
Number of assembled bodies         = 637
Number of uses of BODY HOLES       = 4
Maximum HOLE material used         = 7
Number of uses of each input part:
  20  4  80  16  16  4  4  16  4  4
  1   4   4   4   2   1   1   1   1   1
  1

Scoring region volumes have been multiplied by the number of uses of each part.

External zone of global part is number 123 with definition: -15

The coordinate limits of the geometry are estimated to be:
XMIN= -113.030 XMAX= 113.030
YMIN= -113.030 YMAX= 113.030
ZMIN= -24.130 ZMAX= 390.525

Albedo data for outer surface of global part 21
Body number= 103 Shape number= 5
The material number in the outside zone ( 123 ) has been reset to -4000

Reflection coefficients for 3 surfaces are:
1.0000 1.0000 1.0000
A periodic boundary condition has been requested.

TIME TAKEN TO READ AND PROCESS GEOMETRY DATA= 1.563 SECS
STORAGE USED FOR GEOMETRY DATA (4BYTE WORDS)= 16814
MAXIMUM REAL MATERIAL NUMBER USED = 19
MAXIMUM HOLE MATERIAL REFERENCED IN GEOMETRY= 7

```

```

*****
* SUCCESSFUL CONCLUSION TO UNIT 2 TO READ THE *
* MATERIAL GEOMETRY DATA. NO ERRORS.         *
*****

```

1

```

*****
* ENTERING UNIT 3 TO READ HOLE DATA          *
*****

```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

Changing input to named channel: FHF after reading
189 lines from named channel: INPUT Total line count= 816

HOLE 1 PLATE

DIRECTION COSINES OF PLATE NORMAL = 0.0000 0.0000 1.0000
NO. OF EXPLICIT PLATE BOUNDARIES = 2
The input characters : [@lassem-@ltnozz]
have been replaced by : +3.3088577E+02
The input characters : [@lbnnozz]
have been replaced by : +8.097500

PLATE BOUNDARY	MATERIAL NO.
	8
330.8858	-2
8.0975	7

HOLE 2 LATTICE

The input characters : [-(@rpitch/2.0)-(@wfuel-@lattice*@rpitch)/2.0]
have been replaced by : -.7214997
The input characters : [-(@rpitch/2.0)-(@wfuel-@lattice*@rpitch)/2.0]
have been replaced by : -.7214997
The input characters : [@odgt/2.0]
have been replaced by : +.6731000
The input characters : [@odgt/2.]
have been replaced by : +.6731000

NO. OF PINS IN THE X-DIRECTION = 15 NO. OF PINS IN THE Y-DIRECTION = 15
LATTICE PITCH IN THE X-DIRECTION = 1.4427 LATTICE PITCH IN THE Y-DIRECTION = 1.4427
ORIGIN SHIFT (X0,Y0) = -0.7215 -0.7215

ALL PINS HAVE INNER RADIUS 0.6731 AND OUTER RADIUS 0.6731

PINS 1 TO 32 CONTAIN ROD MATERIAL NO. = -3
PINS 33 TO 33 CONTAIN ROD MATERIAL NO. = -4
PINS 34 TO 35 CONTAIN ROD MATERIAL NO. = -3
PINS 36 TO 36 CONTAIN ROD MATERIAL NO. = -4
PINS 37 TO 39 CONTAIN ROD MATERIAL NO. = -3
PINS 40 TO 40 CONTAIN ROD MATERIAL NO. = -4
PINS 41 TO 42 CONTAIN ROD MATERIAL NO. = -3
PINS 43 TO 43 CONTAIN ROD MATERIAL NO. = -4
PINS 44 TO 52 CONTAIN ROD MATERIAL NO. = -3
PINS 53 TO 53 CONTAIN ROD MATERIAL NO. = -4
PINS 54 TO 64 CONTAIN ROD MATERIAL NO. = -3
PINS 65 TO 65 CONTAIN ROD MATERIAL NO. = -4
PINS 66 TO 70 CONTAIN ROD MATERIAL NO. = -3
PINS 71 TO 71 CONTAIN ROD MATERIAL NO. = -4
PINS 72 TO 77 CONTAIN ROD MATERIAL NO. = -3
PINS 78 TO 78 CONTAIN ROD MATERIAL NO. = -4
PINS 79 TO 87 CONTAIN ROD MATERIAL NO. = -3
PINS 88 TO 88 CONTAIN ROD MATERIAL NO. = -4
PINS 89 TO 108 CONTAIN ROD MATERIAL NO. = -3
PINS 109 TO 109 CONTAIN ROD MATERIAL NO. = -4
PINS 110 TO 112 CONTAIN ROD MATERIAL NO. = -3
PINS 113 TO 113 CONTAIN ROD MATERIAL NO. = -5
PINS 114 TO 116 CONTAIN ROD MATERIAL NO. = -3
PINS 117 TO 117 CONTAIN ROD MATERIAL NO. = -4
PINS 118 TO 137 CONTAIN ROD MATERIAL NO. = -3
PINS 138 TO 138 CONTAIN ROD MATERIAL NO. = -4
PINS 139 TO 147 CONTAIN ROD MATERIAL NO. = -3
PINS 148 TO 148 CONTAIN ROD MATERIAL NO. = -4
PINS 149 TO 154 CONTAIN ROD MATERIAL NO. = -3
PINS 155 TO 155 CONTAIN ROD MATERIAL NO. = -4
PINS 156 TO 160 CONTAIN ROD MATERIAL NO. = -3
PINS 161 TO 161 CONTAIN ROD MATERIAL NO. = -4

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

PINS 162 TO 172 CONTAIN ROD MATERIAL NO. = -3
PINS 173 TO 173 CONTAIN ROD MATERIAL NO. = -4
PINS 174 TO 182 CONTAIN ROD MATERIAL NO. = -3
PINS 183 TO 183 CONTAIN ROD MATERIAL NO. = -4
PINS 184 TO 185 CONTAIN ROD MATERIAL NO. = -3
PINS 186 TO 186 CONTAIN ROD MATERIAL NO. = -4
PINS 187 TO 189 CONTAIN ROD MATERIAL NO. = -3
PINS 190 TO 190 CONTAIN ROD MATERIAL NO. = -4
PINS 191 TO 192 CONTAIN ROD MATERIAL NO. = -3
PINS 193 TO 193 CONTAIN ROD MATERIAL NO. = -4
PINS 194 TO 225 CONTAIN ROD MATERIAL NO. = -3

ALL PINS HAVE CAN MATERIAL NO. = 3

OTHERWISE MATERIAL NO. = 3

HOLE 3 RZMESH

The input characters : [@odpeller/2.0]
have been replaced by : +.4653500
The input characters : [(@odclad/2.0)-@cladth]
have been replaced by : +.4768000
The input characters : [@odclad/2.0]
have been replaced by : +.5334000

3 R-MESHES WITH BOUNDARIES LOCATED AT :

4.65350E-01 4.76800E-01 5.33400E-01
The input characters : [@lbnnozz+@fgapb]
have been replaced by : +8.097500
The input characters : [@lbnnozz+@fgapb+@lbcap]
have been replaced by : +9.837399
The input characters : [@lbnnozz+@fgapb+@lbcap+@lfexpb]
have been replaced by : +9.837399
The input characters : [@lbnnozz+@fgapb+@lbcap+@lfexpb+@lactfu]
have been replaced by : +3.1603439E+02
The input characters : [@lbnnozz+@fgapb+@lfu-@ltcap]
have been replaced by : +3.2822641E+02
The input characters : [@lbnnozz+@fgapb+@lfu]
have been replaced by : +3.2996631E+02
The input characters : [@lassem-@ltnozz]
have been replaced by : +3.3088577E+02

7 Z-MESHES WITH BOUNDARIES LOCATED AT :

8.09750E+00 8.09750E+00 9.83740E+00 9.83740E+00 3.16034E+02 3.28226E+02 3.29966E+02 3.30886E+02

MATERIAL MAP

Z-MESH 1 :
3 3 3

Z-MESH 2 :
2 2 2

Z-MESH 3 :
4 4 2

Z-MESH 4 :
1 4 2

Z-MESH 5 :
4 4 2

Z-MESH 6 :
2 2 2

Z-MESH 7 :
3 3 3

MATERIAL OUTSIDE DEFINED MESHES : 3

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

```
HOLE 4      RZMESH
-----
The input characters      : [(@odgt/2.0) - @gtth]
have been replaced by    : +.6401000
The input characters      : [(@odgt/2.0)]
have been replaced by    : +.6731000

      2 R-MESHERS WITH BOUNDARIES LOCATED AT :
6.40100E-01 6.73100E-01
The input characters      : [(@lassem - @ltnozz)]
have been replaced by    : +3.3088577E+02

      1 Z-MESHERS WITH BOUNDARIES LOCATED AT :
8.09750E+00 3.30886E+02

      MATERIAL MAP
      -----

      Z-MESH  1 :
              3  5

      MATERIAL OUTSIDE DEFINED MESHERS :  3

HOLE 5      RZMESH
-----
The input characters      : [(@odit/2.0) - @itth]
have been replaced by    : +.6401000
The input characters      : [(@odit/2.0)]
have been replaced by    : +.6731000

      2 R-MESHERS WITH BOUNDARIES LOCATED AT :
6.40100E-01 6.73100E-01
The input characters      : [(@lassem - @ltnozz)]
have been replaced by    : +3.3088577E+02

      1 Z-MESHERS WITH BOUNDARIES LOCATED AT :
8.09750E+00 3.30886E+02

      MATERIAL MAP
      -----

      Z-MESH  1 :
              3  6

      MATERIAL OUTSIDE DEFINED MESHERS :  3
Returning input to named channel:  INPUT after reading
93 lines from named channel:      FHF Total line count= 905

Changing input to named channel:  CHF after reading
190 lines from named channel:     INPUT Total line count= 906

HOLE 6      PLATE
-----

      DIRECTION COSINES OF PLATE NORMAL = 1.0000 0.0000 0.0000
      NO. OF EXPLICIT PLATE BOUNDARIES =  2
The input characters      : [(@blt+@blct)/2.0]
have been replaced by    : +.1587500
The input characters      : [(@blt+@blct)/2.0]
have been replaced by    : +.0317500

      PLATE BOUNDARY      MATERIAL NO.

                                11

                        0.1587
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

0.0318 10
11
Returning input to named channel: INPUT after reading
17 lines from named channel: CHF Total line count= 917
Changing input to named channel: CHF after reading
191 lines from named channel: INPUT Total line count= 918

HOLE 7 PLATE

DIRECTION COSINES OF PLATE NORMAL = 0.0000 0.0000 1.0000
NO. OF EXPLICIT PLATE BOUNDARIES = 5
The input characters : [@lbasket-@lbws-@lbw]
have been replaced by : +3.4394141E+02
The input characters : [-1.0*(@lbw+@lbws)]
have been replaced by : -15.46860
The input characters : [-1.0*(@baskoff+@lbw+@lbws)]
have been replaced by : -15.46860

PLATE BOUNDARY	MATERIAL NO.
343.9414	3
316.0522	-8
0.0000	-10
-15.4686	-9
-15.4686	3
	3

HOLE 8 RZMESH

The input characters : [@diatpw/2.0]
have been replaced by : +87.60460

1 R-MESHES WITH BOUNDARIES LOCATED AT :
8.76046E+01
The input characters : [@lbasket-@lbws-@lbw-@ltpw]
have been replaced by : +3.2666940E+02
The input characters : [@lbasket-@lbws-@lbw-@ltpw+@ltpwd]
have been replaced by : +3.2793939E+02

1 Z-MESHES WITH BOUNDARIES LOCATED AT :
3.26669E+02 3.27939E+02

MATERIAL MAP

Z-MESH 1 :
13

MATERIAL OUTSIDE DEFINED MESHES : 3

HOLE 9 RZMESH

The input characters : [@diabw/2.0]
have been replaced by : +87.60460

1 R-MESHES WITH BOUNDARIES LOCATED AT :
8.76046E+01

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
The input characters      : [-1.0*(@lbws+@lbwd)]
have been replaced by    : -11.65860
The input characters      : [-1.0*(@lbws)]
have been replaced by    : -10.38860

      1 Z-MESHES WITH BOUNDARIES LOCATED AT :
-1.16586E+01-1.03886E+01

MATERIAL MAP
-----

Z-MESH  1 :
13

MATERIAL OUTSIDE DEFINED MESHES :  3

HOLE 10   PLATE
-----

      DIRECTION COSINES OF PLATE NORMAL = 0.0000 0.0000 1.0000
      NO. OF EXPLICIT PLATE BOUNDARIES =  4
The input characters      : [@tspd+@dsspht+@thtd+@dsspht]
have been replaced by    : +11.65860

      REPEATED CELL OF WIDTH 1.16586E+01
The input characters      : [@tspd+@dsspht+@thtd+@dsspht]
have been replaced by    : +11.65860
The input characters      : [@tspd+@dsspht+@thtd]
have been replaced by    : +7.099300
The input characters      : [@tspd+@dsspht]
have been replaced by    : +5.829300

      PLATE BOUNDARY      MATERIAL NO.
      -----
      11.6586             3
      7.0993              3
      5.8293              14
      1.2700              3
                           12

HOLE  1 : NO. OF MATERIALS =  8  MATERIAL LIST =  8  7  3  2  4  1  5  6
HOLE  2 : NO. OF MATERIALS =  6  MATERIAL LIST =  3  2  4  1  5  6
HOLE  3 : NO. OF MATERIALS =  4  MATERIAL LIST =  3  2  4  1
HOLE  4 : NO. OF MATERIALS =  2  MATERIAL LIST =  3  5
HOLE  5 : NO. OF MATERIALS =  2  MATERIAL LIST =  3  6
HOLE  6 : NO. OF MATERIALS =  2  MATERIAL LIST = 11 10
HOLE  7 : NO. OF MATERIALS =  4  MATERIAL LIST =  3 13 14 12
HOLE  8 : NO. OF MATERIALS =  2  MATERIAL LIST = 13  3
HOLE  9 : NO. OF MATERIALS =  2  MATERIAL LIST = 13  3
HOLE 10 : NO. OF MATERIALS =  3  MATERIAL LIST =  3 14 12

      Returning input to named channel:      INPUT after reading
      74 lines from named channel:          CHF Total line count= 971

*****
* SUCCESSFUL CONCLUSION TO UNIT 3 TO READ THE *
* HOLE DATA NO ERRORS. *
*****

1

*****
* ENTERING UNIT 4 TO READ PROBLEM CONTROL DATA. *
*****
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
*****
USER SUPPLIED RANDOM NUMBER SEEDS  = 54321
                                     54321

FOR A CRITICALITY CALCULATION :
NUMBER OF SUPER HISTORIES PER BATCH = 1000
FIRST STAGE NUMBER                   = -3
LAST STAGE NUMBER                   = 810
MAXIMUM GENERATION NUMBER           = 10
ESTIMATE OF NUFACTOR                 = 1.000
FINAL STANDARD DEVIATION ON K(THREE) = 0.000800

SAMPLING TIME ( SECONDS )           = 10000000
ERRORS PERMITTING, AN EXECUTION OF THE SUPPLIED
DATA WILL BE ATTEMPTED

*****
* SUCCESSFUL CONCLUSION TO UNIT 4 TO READ THE *
* PROBLEM CONTROL DATA. NO ERRORS.          *
*****

1

*****
* ENTERING UNIT 5 TO READ SOURCE GEOMETRY DATA. *
*****

SOURCE GEOMETRY SUCCESSFULLY INPUT
SOURCE WILL BE SAMPLED FROM 1 REGIONS

*****
* SUCCESSFUL CONCLUSION TO UNIT 5 TO READ THE *
* SOURCE GEOMETRY DATA. NO ERRORS.          *
*****

1

*****QI*****
* STATUS OF INPUT UNITS AT END OF STAGE ONE PROCESSING *
*****

PROBLEM CONTROL DATA.    USED-OK
MATERIAL GEOMETRY DATA.  USED-OK
SOURCE GEOMETRY DATA.    USED-OK
HOLE DATA                USED-OK
MATERIAL SPECIFICATION     USED-OK

*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 25005 (FG TRACKING)                        *
*****

*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 48321 (SCORING)                          *
*****

WARNING: Negative volumes have been found in the following regions:

REGION      VOLUME
-----
11          -1.382E+05
12          -2.649E+06
15          -3.455E+04
16          -6.624E+05
19          -3.848E+04
```

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

20	-7.244E+05
87	-1.859E+05
90	-9.965E+06
93	-8.286E+06
95	-9.863E+06
102	-1.011E+05

Inspect the FG printout to ensure that these do not correspond to scoring regions.

These negative volumes have been reset to 1.0

```
*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 03435 (SIMPLE SOURCE) *
*****
```

SOURCE WILL BE SAMPLED FROM THE FOLLOWING ZONES:

ENVELOPE					
XHI	XLO	YHI	YLO	ZHI	ZLO

RESTRICTION					
FGZONES					
MATERIAL					

1.1303E+02	-1.1303E+02	1.1303E+02	-1.1303E+02	3.9052E+02	-2.4130E+01
108	122				1

```
*****
ANSWERS Software Licensing Procedure for Modules
Module name - MONK-CRIT
*****
```

```
*** The timelock password has been successfully verified.
*** This password expires at midnight on 30/ 6/2001
```

```
*** Successful authorisation achieved on system 57NST
*** This system has the identification number 131008778
```

```
*****
This module is part of the ANSWERS program - MONK8A

The use of this module is restricted within the terms of your
licence agreement with AEA Technology.
*****
```

```
*****
*** End of ANSWERS Software Licensing Checks ***
*****
```

```
*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 36882 (POINT NEUTRON) *
*****
```

1

DICE MK. VII Sept 92 (BOUND-ATOM THERMAL TREATMENT FOR CERTAIN NUCLIDES), LIBRARY FORMAT VERSION 3
=====

NUCLEAR DATA READ FROM FILE: d:\answers\data_libraries\dice96j2v5.dat

THIS CASE IS BEING RUN FOR 20 MATERIAL(S) WITH 13193 CROSS-SECTION ENERGY GROUPS FROM 1.50000E+01 MEV TO 1.00000E-11 MEV

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

THERMAL DATA READ FOR FOLLOWING NUCLIDES							
NUCLIDE ID NO	COMMENTS						
10293	Data for hydrogen in H2O processed from JEF file JAN 92 by D.E.Bendall using 3% fitting error and .1% integration error						
THE THERMAL DATA OCCUPIES 14611 LOCATIONS; THE TOTAL NUCLEAR DATA OCCUPIES 3618571 LOCATIONS; THE DATA TOOK 4.1 SECS OF CPU TO PROCESS							
SUMMARY OF MATERIAL DATA							
MATERIAL 1	DENSITY 1.04120E+01 GRMS/CC	NUMBER OF NUCLIDES 3		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	1	J2U235	92	235.04	1.5546E-02	9228b	62
	2	J2U238	92	238.05	3.1765E-01	9237b	63
	3	J2O16	8	15.99	6.6681E-01	825b	33
MATERIAL 2	DENSITY 6.50000E+00 GRMS/CC	NUMBER OF NUCLIDES 1		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	1	J2ZR	40	91.22	1.0000E+00	4000b	47
MATERIAL 3	DENSITY 9.98200E-01 GRMS/CC	NUMBER OF NUCLIDES 2		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	1	J2HINH2O	1	1.01	6.6661E-01	10293	64
	2	J2O16	8	15.99	3.3339E-01	825b	33
MATERIAL 4	DENSITY 9.98200E-01 GRMS/CC	NUMBER OF NUCLIDES 2		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	1	J2HINH2O	1	1.01	6.6661E-01	10293	64
	2	J2O16	8	15.99	3.3339E-01	825b	33
MATERIAL 5	DENSITY 6.50000E+00 GRMS/CC	NUMBER OF NUCLIDES 1		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	1	J2ZR	40	91.22	1.0000E+00	4000b	47
MATERIAL 6	DENSITY 6.50000E+00 GRMS/CC	NUMBER OF NUCLIDES 1		TEMPERATURE 293.0 DEGREES KELVIN			
	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

1	J2ZR	40	91.22	1.0000E+00	4000b	47
MATERIAL 7	DENSITY 3.09913E+00 GRMS/CC	NUMBER OF NUCLIDES 15	TEMPERATURE 293.0 DEGREES KELVIN			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
1	J2FE54	26	53.94	1.1794E-02	2625b	15
2	J2FE56	26	55.94	1.8334E-01	2631b	66
3	J2FE57	26	56.94	4.1978E-03	2634b	16
4	J2FE58	26	57.93	5.5970E-04	2637b	17
5	J2CR50	24	49.95	2.2691E-03	2425b	11
6	J2CR52	24	51.94	4.3759E-02	2431b	53
7	J2CR53	24	52.94	4.9613E-03	2434b	12
8	J2CR54	24	53.94	1.2351E-03	2437b	13
9	J2NI58	28	57.94	1.3998E-02	2825b	56
10	J2NI60	28	59.93	5.3920E-03	2831b	57
11	J2NI61	28	60.93	2.3441E-04	2834b	58
12	J2NI62	28	61.93	7.4723E-04	2837b	59
13	J2NI64	28	63.93	1.9040E-04	2843b	60
14	J2HINH2O	1	1.01	4.8484E-01	10293	64
15	J2O16	8	15.99	2.4249E-01	825b	33
MATERIAL 8	DENSITY 2.19041E+00 GRMS/CC	NUMBER OF NUCLIDES 15	TEMPERATURE 293.0 DEGREES KELVIN			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
1	J2FE54	26	53.94	6.5686E-03	2625b	15
2	J2FE56	26	55.94	1.0211E-01	2631b	66
3	J2FE57	26	56.94	2.3380E-03	2634b	16
4	J2FE58	26	57.93	3.1172E-04	2637b	17
5	J2CR50	24	49.95	1.2638E-03	2425b	11
6	J2CR52	24	51.94	2.4371E-02	2431b	53
7	J2CR53	24	52.94	2.7632E-03	2434b	12
8	J2CR54	24	53.94	6.8789E-04	2437b	13
9	J2NI58	28	57.94	7.7962E-03	2825b	56
10	J2NI60	28	59.93	3.0031E-03	2831b	57
11	J2NI61	28	60.93	1.3055E-04	2834b	58
12	J2NI62	28	61.93	4.1617E-04	2837b	59
13	J2NI64	28	63.93	1.0605E-04	2843b	60
14	J2HINH2O	1	1.01	5.6537E-01	10293	64
15	J2O16	8	15.99	2.8276E-01	825b	33
MATERIAL 9	DENSITY 7.92967E+00 GRMS/CC	NUMBER OF NUCLIDES 13	TEMPERATURE 293.0 DEGREES KELVIN			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
1	J2FE54	26	53.94	4.3251E-02	2625b	15
2	J2FE56	26	55.94	6.7237E-01	2631b	66
3	J2FE57	26	56.94	1.5395E-02	2634b	16
4	J2FE58	26	57.93	2.0526E-03	2637b	17
5	J2CR50	24	49.95	8.3216E-03	2425b	11
6	J2CR52	24	51.94	1.6048E-01	2431b	53
7	J2CR53	24	52.94	1.8195E-02	2434b	12
8	J2CR54	24	53.94	4.5295E-03	2437b	13
9	J2NI58	28	57.94	5.1335E-02	2825b	56
10	J2NI60	28	59.93	1.9774E-02	2831b	57
11	J2NI61	28	60.93	8.5964E-04	2834b	58
12	J2NI62	28	61.93	2.7403E-03	2837b	59
13	J2NI64	28	63.93	6.9827E-04	2843b	60
MATERIAL 10	DENSITY 2.58326E+00 GRMS/CC	NUMBER OF NUCLIDES 4	TEMPERATURE 293.0 DEGREES KELVIN			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
1	J2AL27	13	26.98	3.6469E-01	1325b	1

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

	2	J2B10	5	10.01	7.7071E-02	525b	5
	3	J2B11	5	11.01	4.2603E-01	528b	48
	4	J2C	6	12.01	1.3221E-01	600b	6

MATERIAL 11	DENSITY 2.70000E+00 GRMS/CC	NUMBER OF NUCLIDES 1	TEMPERATURE 293.0 DEGREES KELVIN			
-----	-----	-----	-----			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	-----	-----
1	J2AL27	13	26.98	1.0000E+00	1325b	1

MATERIAL 12	DENSITY 7.92967E+00 GRMS/CC	NUMBER OF NUCLIDES 13	TEMPERATURE 293.0 DEGREES KELVIN			
-----	-----	-----	-----			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	-----	-----
1	J2FE54	26	53.94	4.3251E-02	2625b	15
2	J2FE56	26	55.94	6.7237E-01	2631b	66
3	J2FE57	26	56.94	1.5395E-02	2634b	16
4	J2FE58	26	57.93	2.0526E-03	2637b	17
5	J2CR50	24	49.95	8.3216E-03	2425b	11
6	J2CR52	24	51.94	1.6048E-01	2431b	53
7	J2CR53	24	52.94	1.8195E-02	2434b	12
8	J2CR54	24	53.94	4.5295E-03	2437b	13
9	J2NI58	28	57.94	5.1335E-02	2825b	56
10	J2NI60	28	59.93	1.9774E-02	2831b	57
11	J2NI61	28	60.93	8.5964E-04	2834b	58
12	J2NI62	28	61.93	2.7403E-03	2837b	59
13	J2NI64	28	63.93	6.9827E-04	2843b	60

MATERIAL 13	DENSITY 7.92967E+00 GRMS/CC	NUMBER OF NUCLIDES 13	TEMPERATURE 293.0 DEGREES KELVIN			
-----	-----	-----	-----			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	-----	-----
1	J2FE54	26	53.94	4.3251E-02	2625b	15
2	J2FE56	26	55.94	6.7237E-01	2631b	66
3	J2FE57	26	56.94	1.5395E-02	2634b	16
4	J2FE58	26	57.93	2.0526E-03	2637b	17
5	J2CR50	24	49.95	8.3216E-03	2425b	11
6	J2CR52	24	51.94	1.6048E-01	2431b	53
7	J2CR53	24	52.94	1.8195E-02	2434b	12
8	J2CR54	24	53.94	4.5295E-03	2437b	13
9	J2NI58	28	57.94	5.1335E-02	2825b	56
10	J2NI60	28	59.93	1.9774E-02	2831b	57
11	J2NI61	28	60.93	8.5964E-04	2834b	58
12	J2NI62	28	61.93	2.7403E-03	2837b	59
13	J2NI64	28	63.93	6.9827E-04	2843b	60

MATERIAL 14	DENSITY 2.70000E+00 GRMS/CC	NUMBER OF NUCLIDES 1	TEMPERATURE 293.0 DEGREES KELVIN			
-----	-----	-----	-----			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	-----	-----
1	J2AL27	13	26.98	1.0000E+00	1325b	1

MATERIAL 15	DENSITY 7.92967E+00 GRMS/CC	NUMBER OF NUCLIDES 13	TEMPERATURE 293.0 DEGREES KELVIN			
-----	-----	-----	-----			
NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	-----	-----
1	J2FE54	26	53.94	4.3251E-02	2625b	15
2	J2FE56	26	55.94	6.7237E-01	2631b	66
3	J2FE57	26	56.94	1.5395E-02	2634b	16
4	J2FE58	26	57.93	2.0526E-03	2637b	17
5	J2CR50	24	49.95	8.3216E-03	2425b	11
6	J2CR52	24	51.94	1.6048E-01	2431b	53
7	J2CR53	24	52.94	1.8195E-02	2434b	12
8	J2CR54	24	53.94	4.5295E-03	2437b	13
9	J2NI58	28	57.94	5.1335E-02	2825b	56

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

10	J2NI60	28	59.93	1.9774E-02	2831b	57
11	J2NI61	28	60.93	8.5964E-04	2834b	58
12	J2NI62	28	61.93	2.7403E-03	2837b	59
13	J2NI64	28	63.93	6.9827E-04	2843b	60

MATERIAL 16	DENSITY 7.92967E+00 GRMS/CC		NUMBER OF NUCLIDES 13		TEMPERATURE 293.0 DEGREES KELVIN		

	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	-----	----	-----	-----	-----	---	-----
	1	J2FE54	26	53.94	4.3251E-02	2625b	15
	2	J2FE56	26	55.94	6.7237E-01	2631b	66
	3	J2FE57	26	56.94	1.5395E-02	2634b	16
	4	J2FE58	26	57.93	2.0526E-03	2637b	17
	5	J2CR50	24	49.95	8.3216E-03	2425b	11
	6	J2CR52	24	51.94	1.6048E-01	2431b	53
	7	J2CR53	24	52.94	1.8195E-02	2434b	12
	8	J2CR54	24	53.94	4.5295E-03	2437b	13
	9	J2NI58	28	57.94	5.1335E-02	2825b	56
	10	J2NI60	28	59.93	1.9774E-02	2831b	57
	11	J2NI61	28	60.93	8.5964E-04	2834b	58
	12	J2NI62	28	61.93	2.7403E-03	2837b	59
	13	J2NI64	28	63.93	6.9827E-04	2843b	60

MATERIAL 17	DENSITY 1.10400E+01 GRMS/CC	NUMBER OF NUCLIDES 1	TEMPERATURE 293.0 DEGREES KELVIN			

NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	---	-----	-----	-----	---	-----
1	J2PB	82	207.21	1.0000E+00	8200b	34

MATERIAL 18	DENSITY 1.63187E+00 GRMS/CC		NUMBER OF NUCLIDES 8		TEMPERATURE 293.0 DEGREES KELVIN		

	NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
	-----	----	-----	-----	-----	---	-----
	1	J2B10	5	10.01	7.3169E-04	525b	5
	2	J2B11	5	11.01	2.9274E-03	528b	48
	3	J2AL27	13	26.98	6.6692E-02	1325b	1
	4	J2HINH20	1	1.01	5.0079E-01	10293	64
	5	J2O16	8	15.99	2.2325E-01	825b	33
	6	J2C	6	12.01	1.9368E-01	600b	6
	7	J2NI14	7	14.00	1.1881E-02	725b	31
	8	J2NI15	7	15.00	4.4123E-05	728b	32

MATERIAL 19	DENSITY 9.98200E-01 GRMS/CC	NUMBER OF NUCLIDES 2	TEMPERATURE 293.0 DEGREES KELVIN			

NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	---	-----
1	J2HINH2O	1	1.01	6.6661E-01	10293	64
2	J2O16	8	15.99	3.3339E-01	825b	33

MATERIAL 20	DENSITY 9.98200E-01 GRMS/CC	NUMBER OF NUCLIDES 2	TEMPERATURE 293.0 DEGREES KELVIN			

NUCLIDE NO.	NAME	ATOMIC NO.	ATOMIC WT	PROP BY NUCLIDE	DFN	LIBRARY NO.
-----	-----	-----	-----	-----	---	-----
1	J2HINH2O	1	1.01	6.6661E-01	10293	64
2	J2O16	8	15.99	3.3339E-01	825b	33

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

```
*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 37371 (HOLE MODULE) *
*****
```

```
NO. OF HOLES REFERENCED IN THE GEOMETRY DATA = 7
TOTAL NUMBER OF HOLES = 10
TOTAL NUMBER OF MATERIALS = 19
```

```
*****
* ENTERING ROUTINE TO DISTRIBUTE THE INPUT DATA TO *
* MODULE 16724 (ACTION TALLIES) *
*****
```

```
THE FOLLOWING SCORES WILL BE TALLIED AND PRINTED:
SOURCE DISTRIBUTION COUNTS
BOUNDARY CROSSING COUNTS
MATERIAL ACTION COUNTS
REGION ACTION COUNTS
NEUTRON PARAMETERS
SYSTEM CATEGORISATION
FLUXES
SAMPLING GUIDANCE
ACTION TALLIES WILL BE NORMALISED TO 10000 SAMPLES
```

```
*****Q2*****
* END OF STAGE TWO PROCESSING *
*****
```

AT THE START OF THE TRACKING PROCESS THERE IS OVER A WEEK OF CPU TIME AVAILABLE

A TIME OF 30.000 SECS WILL BE ALLOWED FOR CREATING DUMP AND OUTPUT FILES

STAGE -3 COMPLETED AFTER	0.31 MIN - K(THREE)= 0.9072 (0.0106)
STAGE -2 COMPLETED AFTER	0.69 MIN - K(THREE)= 0.9167 (0.0068)
STAGE -1 COMPLETED AFTER	1.05 MIN - K(THREE)= 0.9240 (0.0054)
STAGE 0 COMPLETED AFTER	1.42 MIN - K(THREE)= 0.9222 (0.0048)
STAGE 1 COMPLETED AFTER	1.83 MIN - K(THREE)= 0.9087 (0.0096)
STAGE 2 COMPLETED AFTER	2.28 MIN - K(THREE)= 0.9278 (0.0063)
STAGE 3 COMPLETED AFTER	2.73 MIN - K(THREE)= 0.9303 (0.0052)
STAGE 4 COMPLETED AFTER	3.15 MIN - K(THREE)= 0.9329 (0.0047)
STAGE 5 COMPLETED AFTER	3.54 MIN - K(THREE)= 0.9295 (0.0042)
STAGE 6 COMPLETED AFTER	3.98 MIN - K(THREE)= 0.9308 (0.0039)
STAGE 7 COMPLETED AFTER	4.41 MIN - K(THREE)= 0.9309 (0.0036)
STAGE 8 COMPLETED AFTER	4.82 MIN - K(THREE)= 0.9316 (0.0034)
STAGE 9 COMPLETED AFTER	5.23 MIN - K(THREE)= 0.9308 (0.0032)
STAGE 10 COMPLETED AFTER	5.62 MIN - K(THREE)= 0.9291 (0.0030)
STAGE 11 COMPLETED AFTER	6.05 MIN - K(THREE)= 0.9295 (0.0029)
STAGE 12 COMPLETED AFTER	6.47 MIN - K(THREE)= 0.9289 (0.0028)
STAGE 13 COMPLETED AFTER	6.86 MIN - K(THREE)= 0.9279 (0.0027)
STAGE 14 COMPLETED AFTER	7.26 MIN - K(THREE)= 0.9279 (0.0025)
STAGE 15 COMPLETED AFTER	7.65 MIN - K(THREE)= 0.9271 (0.0025)
STAGE 16 COMPLETED AFTER	8.13 MIN - K(THREE)= 0.9280 (0.0024)
STAGE 17 COMPLETED AFTER	8.60 MIN - K(THREE)= 0.9286 (0.0023)
STAGE 18 COMPLETED AFTER	9.02 MIN - K(THREE)= 0.9284 (0.0022)
STAGE 19 COMPLETED AFTER	9.54 MIN - K(THREE)= 0.9298 (0.0022)
STAGE 20 COMPLETED AFTER	9.98 MIN - K(THREE)= 0.9300 (0.0021)
STAGE 21 COMPLETED AFTER	10.43 MIN - K(THREE)= 0.9302 (0.0020)
STAGE 22 COMPLETED AFTER	10.89 MIN - K(THREE)= 0.9305 (0.0020)
STAGE 23 COMPLETED AFTER	11.33 MIN - K(THREE)= 0.9309 (0.0020)
STAGE 24 COMPLETED AFTER	11.75 MIN - K(THREE)= 0.9314 (0.0019)
STAGE 25 COMPLETED AFTER	12.20 MIN - K(THREE)= 0.9315 (0.0019)
STAGE 26 COMPLETED AFTER	12.61 MIN - K(THREE)= 0.9313 (0.0019)

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

STAGE	27	COMPLETED AFTER	13.03 MIN - K(THREE) = 0.9310 (0.0018)
STAGE	28	COMPLETED AFTER	13.44 MIN - K(THREE) = 0.9311 (0.0018)
STAGE	29	COMPLETED AFTER	13.85 MIN - K(THREE) = 0.9311 (0.0018)
STAGE	30	COMPLETED AFTER	14.32 MIN - K(THREE) = 0.9311 (0.0017)
STAGE	31	COMPLETED AFTER	14.78 MIN - K(THREE) = 0.9315 (0.0017)
STAGE	32	COMPLETED AFTER	15.18 MIN - K(THREE) = 0.9317 (0.0017)
STAGE	33	COMPLETED AFTER	15.61 MIN - K(THREE) = 0.9319 (0.0017)
STAGE	34	COMPLETED AFTER	16.03 MIN - K(THREE) = 0.9319 (0.0016)
STAGE	35	COMPLETED AFTER	16.44 MIN - K(THREE) = 0.9316 (0.0016)
STAGE	36	COMPLETED AFTER	16.89 MIN - K(THREE) = 0.9318 (0.0016)
STAGE	37	COMPLETED AFTER	17.40 MIN - K(THREE) = 0.9320 (0.0016)
STAGE	38	COMPLETED AFTER	17.82 MIN - K(THREE) = 0.9316 (0.0015)
STAGE	39	COMPLETED AFTER	18.28 MIN - K(THREE) = 0.9318 (0.0015)
STAGE	40	COMPLETED AFTER	18.74 MIN - K(THREE) = 0.9315 (0.0015)
STAGE	41	COMPLETED AFTER	19.28 MIN - K(THREE) = 0.9319 (0.0015)
STAGE	42	COMPLETED AFTER	19.76 MIN - K(THREE) = 0.9321 (0.0015)
STAGE	43	COMPLETED AFTER	20.22 MIN - K(THREE) = 0.9318 (0.0014)
STAGE	44	COMPLETED AFTER	20.75 MIN - K(THREE) = 0.9320 (0.0014)
STAGE	45	COMPLETED AFTER	21.24 MIN - K(THREE) = 0.9322 (0.0014)
STAGE	46	COMPLETED AFTER	21.67 MIN - K(THREE) = 0.9321 (0.0014)
STAGE	47	COMPLETED AFTER	22.11 MIN - K(THREE) = 0.9319 (0.0014)
STAGE	48	COMPLETED AFTER	22.54 MIN - K(THREE) = 0.9317 (0.0014)
STAGE	49	COMPLETED AFTER	23.01 MIN - K(THREE) = 0.9316 (0.0013)
STAGE	50	COMPLETED AFTER	23.51 MIN - K(THREE) = 0.9320 (0.0013)
STAGE	51	COMPLETED AFTER	23.95 MIN - K(THREE) = 0.9320 (0.0013)
STAGE	52	COMPLETED AFTER	24.45 MIN - K(THREE) = 0.9322 (0.0013)
STAGE	53	COMPLETED AFTER	24.91 MIN - K(THREE) = 0.9322 (0.0013)
STAGE	54	COMPLETED AFTER	25.35 MIN - K(THREE) = 0.9320 (0.0013)
STAGE	55	COMPLETED AFTER	25.81 MIN - K(THREE) = 0.9320 (0.0013)
STAGE	56	COMPLETED AFTER	26.24 MIN - K(THREE) = 0.9318 (0.0013)
STAGE	57	COMPLETED AFTER	26.77 MIN - K(THREE) = 0.9321 (0.0012)
STAGE	58	COMPLETED AFTER	27.32 MIN - K(THREE) = 0.9324 (0.0012)
STAGE	59	COMPLETED AFTER	27.78 MIN - K(THREE) = 0.9324 (0.0012)
STAGE	60	COMPLETED AFTER	28.27 MIN - K(THREE) = 0.9326 (0.0012)
STAGE	61	COMPLETED AFTER	28.73 MIN - K(THREE) = 0.9326 (0.0012)
STAGE	62	COMPLETED AFTER	29.18 MIN - K(THREE) = 0.9325 (0.0012)
STAGE	63	COMPLETED AFTER	29.65 MIN - K(THREE) = 0.9324 (0.0012)
STAGE	64	COMPLETED AFTER	30.11 MIN - K(THREE) = 0.9323 (0.0012)
STAGE	65	COMPLETED AFTER	30.59 MIN - K(THREE) = 0.9323 (0.0012)
STAGE	66	COMPLETED AFTER	31.06 MIN - K(THREE) = 0.9323 (0.0012)
STAGE	67	COMPLETED AFTER	31.55 MIN - K(THREE) = 0.9324 (0.0011)
STAGE	68	COMPLETED AFTER	32.02 MIN - K(THREE) = 0.9326 (0.0011)
STAGE	69	COMPLETED AFTER	32.48 MIN - K(THREE) = 0.9326 (0.0011)
STAGE	70	COMPLETED AFTER	32.92 MIN - K(THREE) = 0.9328 (0.0011)
STAGE	71	COMPLETED AFTER	33.36 MIN - K(THREE) = 0.9328 (0.0011)
STAGE	72	COMPLETED AFTER	33.83 MIN - K(THREE) = 0.9330 (0.0011)
STAGE	73	COMPLETED AFTER	34.25 MIN - K(THREE) = 0.9327 (0.0011)
STAGE	74	COMPLETED AFTER	34.69 MIN - K(THREE) = 0.9326 (0.0011)
STAGE	75	COMPLETED AFTER	35.16 MIN - K(THREE) = 0.9327 (0.0011)
STAGE	76	COMPLETED AFTER	35.64 MIN - K(THREE) = 0.9327 (0.0011)
STAGE	77	COMPLETED AFTER	36.07 MIN - K(THREE) = 0.9325 (0.0011)
STAGE	78	COMPLETED AFTER	36.56 MIN - K(THREE) = 0.9325 (0.0011)
STAGE	79	COMPLETED AFTER	37.02 MIN - K(THREE) = 0.9324 (0.0011)
STAGE	80	COMPLETED AFTER	37.50 MIN - K(THREE) = 0.9326 (0.0011)
STAGE	81	COMPLETED AFTER	38.00 MIN - K(THREE) = 0.9326 (0.0010)
STAGE	82	COMPLETED AFTER	38.42 MIN - K(THREE) = 0.9327 (0.0010)
STAGE	83	COMPLETED AFTER	38.90 MIN - K(THREE) = 0.9327 (0.0010)
STAGE	84	COMPLETED AFTER	39.36 MIN - K(THREE) = 0.9326 (0.0010)
STAGE	85	COMPLETED AFTER	39.85 MIN - K(THREE) = 0.9326 (0.0010)
STAGE	86	COMPLETED AFTER	40.28 MIN - K(THREE) = 0.9324 (0.0010)
STAGE	87	COMPLETED AFTER	40.70 MIN - K(THREE) = 0.9323 (0.0010)
STAGE	88	COMPLETED AFTER	41.16 MIN - K(THREE) = 0.9322 (0.0010)
STAGE	89	COMPLETED AFTER	41.62 MIN - K(THREE) = 0.9321 (0.0010)
STAGE	90	COMPLETED AFTER	42.06 MIN - K(THREE) = 0.9320 (0.0010)
STAGE	91	COMPLETED AFTER	42.52 MIN - K(THREE) = 0.9321 (0.0010)
STAGE	92	COMPLETED AFTER	42.97 MIN - K(THREE) = 0.9320 (0.0010)
STAGE	93	COMPLETED AFTER	43.47 MIN - K(THREE) = 0.9321 (0.0010)
STAGE	94	COMPLETED AFTER	43.94 MIN - K(THREE) = 0.9322 (0.0010)
STAGE	95	COMPLETED AFTER	44.41 MIN - K(THREE) = 0.9322 (0.0010)
STAGE	96	COMPLETED AFTER	44.84 MIN - K(THREE) = 0.9322 (0.0010)
STAGE	97	COMPLETED AFTER	45.25 MIN - K(THREE) = 0.9321 (0.0010)
STAGE	98	COMPLETED AFTER	45.75 MIN - K(THREE) = 0.9322 (0.0010)
STAGE	99	COMPLETED AFTER	46.21 MIN - K(THREE) = 0.9320 (0.0010)
STAGE	100	COMPLETED AFTER	46.75 MIN - K(THREE) = 0.9322 (0.0009)

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)

STAGE 101 COMPLETED AFTER	47.24 MIN - K(THREE)= 0.9321 (0.0009)
STAGE 102 COMPLETED AFTER	47.76 MIN - K(THREE)= 0.9319 (0.0009)
STAGE 103 COMPLETED AFTER	48.20 MIN - K(THREE)= 0.9319 (0.0009)
STAGE 104 COMPLETED AFTER	48.65 MIN - K(THREE)= 0.9319 (0.0009)
STAGE 105 COMPLETED AFTER	49.12 MIN - K(THREE)= 0.9319 (0.0009)
STAGE 106 COMPLETED AFTER	49.52 MIN - K(THREE)= 0.9316 (0.0009)
STAGE 107 COMPLETED AFTER	50.02 MIN - K(THREE)= 0.9317 (0.0009)
STAGE 108 COMPLETED AFTER	50.43 MIN - K(THREE)= 0.9316 (0.0009)
STAGE 109 COMPLETED AFTER	50.88 MIN - K(THREE)= 0.9315 (0.0009)
STAGE 110 COMPLETED AFTER	51.35 MIN - K(THREE)= 0.9315 (0.0009)
STAGE 111 COMPLETED AFTER	51.80 MIN - K(THREE)= 0.9314 (0.0009)
STAGE 112 COMPLETED AFTER	52.26 MIN - K(THREE)= 0.9313 (0.0009)
STAGE 113 COMPLETED AFTER	52.71 MIN - K(THREE)= 0.9311 (0.0009)
STAGE 114 COMPLETED AFTER	53.14 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 115 COMPLETED AFTER	53.63 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 116 COMPLETED AFTER	54.08 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 117 COMPLETED AFTER	54.52 MIN - K(THREE)= 0.9309 (0.0009)
STAGE 118 COMPLETED AFTER	54.99 MIN - K(THREE)= 0.9309 (0.0009)
STAGE 119 COMPLETED AFTER	55.47 MIN - K(THREE)= 0.9309 (0.0009)
STAGE 120 COMPLETED AFTER	55.92 MIN - K(THREE)= 0.9309 (0.0009)
STAGE 121 COMPLETED AFTER	56.42 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 122 COMPLETED AFTER	56.89 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 123 COMPLETED AFTER	57.32 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 124 COMPLETED AFTER	57.77 MIN - K(THREE)= 0.9310 (0.0009)
STAGE 125 COMPLETED AFTER	58.23 MIN - K(THREE)= 0.9311 (0.0009)
STAGE 126 COMPLETED AFTER	58.72 MIN - K(THREE)= 0.9311 (0.0008)
STAGE 127 COMPLETED AFTER	59.18 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 128 COMPLETED AFTER	59.61 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 129 COMPLETED AFTER	60.06 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 130 COMPLETED AFTER	60.56 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 131 COMPLETED AFTER	61.03 MIN - K(THREE)= 0.9311 (0.0008)
STAGE 132 COMPLETED AFTER	61.49 MIN - K(THREE)= 0.9311 (0.0008)
STAGE 133 COMPLETED AFTER	61.94 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 134 COMPLETED AFTER	62.40 MIN - K(THREE)= 0.9309 (0.0008)
STAGE 135 COMPLETED AFTER	62.90 MIN - K(THREE)= 0.9309 (0.0008)
STAGE 136 COMPLETED AFTER	63.43 MIN - K(THREE)= 0.9310 (0.0008)
STAGE 137 COMPLETED AFTER	63.94 MIN - K(THREE)= 0.9312 (0.0008)
STAGE 138 COMPLETED AFTER	64.47 MIN - K(THREE)= 0.9312 (0.0008)
STAGE 139 COMPLETED AFTER	64.97 MIN - K(THREE)= 0.9313 (0.0008)
STAGE 140 COMPLETED AFTER	65.39 MIN - K(THREE)= 0.9313 (0.0008)
STAGE 141 COMPLETED AFTER	65.84 MIN - K(THREE)= 0.9313 (0.0008)
STAGE 142 COMPLETED AFTER	66.30 MIN - K(THREE)= 0.9313 (0.0008)

* REQUIRED ACCURACY (SD= 0.000800) REACHED BY K(THREE) *
* EXECUTION ENDS AFTER STAGE 142 *

OUTPUT FROM MONK8A (1/2/1998) AT 18.40.37 ON 15/ 9/2000
=====QO=====

NUMBER OF SUPERHISTORIES POST-SETTLING	=	142000
NUMBER OF SAMPLES POST-SETTLING	=	1426292
FIRST STAGE RUN	=	-3
FINAL STAGE COMPLETED	=	142
TOTAL C.P.U. TIME (SECONDS)	=	3977.996
AVERAGE TIME PER SAMPLE(SECONDS)	=	0.027
AVERAGE TIME PER STAGE(SECONDS)	=	27.247
AVERAGE SAMPLES PER SECONDS	=	36
NUMBER OF RANDOM NUMBERS USED	=	569634827
AFTER SEEDING WITH VALUES	=	54321 54321
TRAJECTORIES TRACKED THIS RUN	=	58521395
NUMBER OF COLLISIONS THIS RUN	=	75810555
ZONE SEARCH LOCATIONS USED(PAIRS)	=	350

***** CUMULATIVE K-EFFECTIVE ESTIMATORS *****											
* STAGE	K(COLL)	STDV	K(SCORE)	STDV	A(SCORE)	STDV	ALPHA	BETA	K(THREE)	STDV	*
* 1	0.9195	0.0129	0.9050	0.0124	0.9973	0.0077	0.0841	0.9162	0.9087	0.0096	*
* 2	0.9291	0.0086	0.9292	0.0088	1.0014	0.0052	0.1753	0.9592	0.9278	0.0063	*
* 3	0.9334	0.0068	0.9298	0.0074	1.0003	0.0044	0.2089	0.9211	0.9303	0.0052	*
* 4	0.9352	0.0060	0.9307	0.0065	0.9985	0.0038	0.2063	0.9112	0.9329	0.0047	*

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)										
*	5	0.9306	0.0054	0.9263	0.0060	0.9975	0.0034	0.2220	0.9323	0.9295 0.0042 *
*	6	0.9319	0.0049	0.9287	0.0056	0.9985	0.0032	0.2460	0.9143	0.9308 0.0039 *
*	7	0.9315	0.0045	0.9285	0.0052	0.9982	0.0030	0.2426	0.9183	0.9309 0.0036 *
*	8	0.9310	0.0043	0.9298	0.0049	0.9984	0.0028	0.2473	0.9013	0.9316 0.0034 *
*	9	0.9288	0.0040	0.9300	0.0045	0.9987	0.0027	0.2675	0.8580	0.9308 0.0032 *
*	10	0.9277	0.0039	0.9277	0.0043	0.9983	0.0026	0.2520	0.8649	0.9291 0.0030 *
*	11	0.9286	0.0037	0.9282	0.0041	0.9987	0.0025	0.2552	0.8648	0.9295 0.0029 *
*	12	0.9290	0.0035	0.9269	0.0040	0.9983	0.0024	0.2499	0.8760	0.9289 0.0028 *
*	13	0.9281	0.0034	0.9267	0.0038	0.9990	0.0023	0.2468	0.8830	0.9279 0.0027 *
*	14	0.9276	0.0033	0.9288	0.0037	1.0007	0.0022	0.2587	0.8642	0.9279 0.0025 *
*	15	0.9270	0.0032	0.9280	0.0036	1.0007	0.0021	0.2613	0.8652	0.9271 0.0025 *
*	16	0.9286	0.0030	0.9285	0.0034	1.0006	0.0020	0.2649	0.8676	0.9280 0.0024 *
*	17	0.9293	0.0029	0.9292	0.0033	1.0008	0.0020	0.2717	0.8546	0.9286 0.0023 *
*	18	0.9291	0.0029	0.9293	0.0032	1.0010	0.0019	0.2718	0.8564	0.9284 0.0022 *
*	19	0.9304	0.0028	0.9308	0.0031	1.0010	0.0018	0.2718	0.8519	0.9298 0.0022 *
*	20	0.9303	0.0027	0.9309	0.0030	1.0009	0.0018	0.2773	0.8525	0.9300 0.0021 *
*	21	0.9305	0.0026	0.9308	0.0030	1.0006	0.0017	0.2762	0.8594	0.9302 0.0020 *
*	22	0.9312	0.0026	0.9312	0.0029	1.0008	0.0017	0.2659	0.8723	0.9305 0.0020 *
*	23	0.9319	0.0025	0.9309	0.0028	1.0003	0.0017	0.2538	0.8826	0.9309 0.0020 *
*	24	0.9320	0.0025	0.9318	0.0028	1.0004	0.0016	0.2604	0.8793	0.9314 0.0019 *
*	25	0.9324	0.0024	0.9319	0.0027	1.0006	0.0016	0.2609	0.8816	0.9315 0.0019 *
*	26	0.9320	0.0024	0.9319	0.0027	1.0007	0.0016	0.2635	0.8757	0.9313 0.0019 *
*	27	0.9317	0.0023	0.9319	0.0026	1.0010	0.0015	0.2681	0.8707	0.9310 0.0018 *
*	28	0.9317	0.0023	0.9314	0.0026	1.0004	0.0015	0.2700	0.8676	0.9311 0.0018 *
*	29	0.9315	0.0022	0.9313	0.0025	1.0004	0.0015	0.2730	0.8701	0.9311 0.0018 *
*	30	0.9319	0.0022	0.9310	0.0025	1.0001	0.0014	0.2682	0.8695	0.9311 0.0017 *
*	31	0.9325	0.0022	0.9318	0.0024	1.0006	0.0014	0.2705	0.8697	0.9315 0.0017 *
*	32	0.9324	0.0021	0.9321	0.0024	1.0005	0.0014	0.2685	0.8785	0.9317 0.0017 *
*	33	0.9325	0.0021	0.9323	0.0024	1.0004	0.0014	0.2634	0.8817	0.9319 0.0017 *
*	34	0.9326	0.0021	0.9321	0.0023	1.0004	0.0014	0.2595	0.8862	0.9319 0.0016 *
*	35	0.9325	0.0020	0.9323	0.0023	1.0008	0.0013	0.2596	0.8896	0.9316 0.0016 *
*	36	0.9327	0.0020	0.9322	0.0023	1.0006	0.0013	0.2576	0.8974	0.9318 0.0016 *
*	37	0.9334	0.0020	0.9327	0.0022	1.0010	0.0013	0.2574	0.8996	0.9320 0.0016 *
*	38	0.9329	0.0020	0.9323	0.0022	1.0010	0.0013	0.2579	0.8990	0.9316 0.0015 *
*	39	0.9330	0.0019	0.9323	0.0022	1.0008	0.0013	0.2579	0.8980	0.9318 0.0015 *
*	40	0.9326	0.0019	0.9320	0.0022	1.0007	0.0013	0.2617	0.8943	0.9315 0.0015 *
*	41	0.9331	0.0019	0.9328	0.0021	1.0012	0.0012	0.2630	0.8960	0.9319 0.0015 *
*	42	0.9333	0.0019	0.9331	0.0021	1.0011	0.0012	0.2665	0.8916	0.9321 0.0015 *
*	43	0.9331	0.0018	0.9326	0.0021	1.0010	0.0012	0.2654	0.8908	0.9318 0.0014 *
*	44	0.9334	0.0018	0.9330	0.0021	1.0012	0.0012	0.2649	0.8900	0.9320 0.0014 *
*	45	0.9337	0.0018	0.9331	0.0020	1.0012	0.0012	0.2643	0.8919	0.9322 0.0014 *
*	46	0.9332	0.0018	0.9330	0.0020	1.0011	0.0012	0.2651	0.8927	0.9321 0.0014 *
*	47	0.9330	0.0018	0.9331	0.0020	1.0013	0.0012	0.2635	0.8964	0.9319 0.0014 *
*	48	0.9327	0.0017	0.9329	0.0020	1.0013	0.0011	0.2621	0.8966	0.9317 0.0014 *
*	49	0.9329	0.0017	0.9331	0.0020	1.0015	0.0011	0.2630	0.8949	0.9316 0.0013 *
*	50	0.9332	0.0017	0.9333	0.0019	1.0015	0.0011	0.2629	0.8946	0.9320 0.0013 *
*	51	0.9332	0.0017	0.9335	0.0019	1.0016	0.0011	0.2630	0.8947	0.9320 0.0013 *
*	52	0.9335	0.0017	0.9338	0.0019	1.0017	0.0011	0.2640	0.8956	0.9322 0.0013 *
*	53	0.9334	0.0017	0.9336	0.0019	1.0015	0.0011	0.2635	0.8952	0.9322 0.0013 *
*	54	0.9332	0.0016	0.9332	0.0019	1.0013	0.0011	0.2644	0.8939	0.9320 0.0013 *
*	55	0.9332	0.0016	0.9330	0.0019	1.0011	0.0011	0.2623	0.8962	0.9320 0.0013 *
*	56	0.9330	0.0016	0.9326	0.0018	1.0009	0.0011	0.2621	0.8976	0.9318 0.0013 *
*	57	0.9331	0.0016	0.9329	0.0018	1.0010	0.0011	0.2645	0.8948	0.9321 0.0012 *
*	58	0.9337	0.0016	0.9333	0.0018	1.0011	0.0011	0.2625	0.8972	0.9324 0.0012 *
*	59	0.9335	0.0016	0.9330	0.0018	1.0008	0.0010	0.2640	0.8949	0.9324 0.0012 *
*	60	0.9339	0.0016	0.9333	0.0018	1.0010	0.0010	0.2622	0.8942	0.9326 0.0012 *
*	61	0.9338	0.0015	0.9333	0.0018	1.0009	0.0010	0.2616	0.8951	0.9326 0.0012 *
*	62	0.9337	0.0015	0.9332	0.0017	1.0009	0.0010	0.2615	0.8935	0.9325 0.0012 *
*	63	0.9338	0.0015	0.9331	0.0017	1.0010	0.0010	0.2614	0.8938	0.9324 0.0012 *
*	64	0.9336	0.0015	0.9334	0.0017	1.0013	0.0010	0.2650	0.8914	0.9323 0.0012 *
*	65	0.9335	0.0015	0.9331	0.0017	1.0011	0.0010	0.2622	0.8940	0.9323 0.0012 *
*	66	0.9336	0.0015	0.9331	0.0017	1.0010	0.0010	0.2599	0.8965	0.9323 0.0012 *
*	67	0.9338	0.0015	0.9333	0.0017	1.0012	0.0010	0.2592	0.8984	0.9324 0.0011 *
*	68	0.9340	0.0015	0.9335	0.0017	1.0012	0.0010	0.2604	0.8981	0.9326 0.0011 *
*	69	0.9339	0.0015	0.9334	0.0017	1.0010	0.0010	0.2611	0.8974	0.9326 0.0011 *
*	70	0.9341	0.0014	0.9333	0.0016	1.0008	0.0010	0.2565	0.9018	0.9328 0.0011 *
*	71	0.9339	0.0014	0.9336	0.0016	1.0010	0.0010	0.2567	0.9052	0.9328 0.0011 *
*	72	0.9341	0.0014	0.9337	0.0016	1.0009	0.0010	0.2537	0.9092	0.9330 0.0011 *
*	73	0.9338	0.0014	0.9333	0.0016	1.0007	0.0009	0.2533	0.9070	0.9327 0.0011 *
*	74	0.9337	0.0014	0.9330	0.0016	1.0006	0.0009	0.2533	0.9056	0.9326 0.0011 *
*	75	0.9338	0.0014	0.9332	0.0016	1.0008	0.0009	0.2546	0.9057	0.9327 0.0011 *
*	76	0.9339	0.0014	0.9333	0.0016	1.0009	0.0009	0.2551	0.9055	0.9327 0.0011 *
*	77	0.9335	0.0014	0.9332	0.0016	1.0009	0.0009	0.2577	0.9037	0.9325 0.0011 *
*	78	0.9336	0.0014	0.9333	0.0016	1.0009	0.0009	0.2576	0.9038	0.9325 0.0011 *

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

* 79	0.9335	0.0014	0.9334	0.0016	1.0011	0.0009	0.2589	0.9023	0.9324	0.0011	*
* 80	0.9336	0.0014	0.9334	0.0015	1.0009	0.0009	0.2587	0.9021	0.9326	0.0011	*
* 81	0.9336	0.0013	0.9334	0.0015	1.0010	0.0009	0.2594	0.9025	0.9326	0.0010	*
* 82	0.9336	0.0013	0.9335	0.0015	1.0009	0.0009	0.2589	0.9052	0.9327	0.0010	*
* 83	0.9337	0.0013	0.9333	0.0015	1.0008	0.0009	0.2555	0.9085	0.9327	0.0010	*
* 84	0.9335	0.0013	0.9332	0.0015	1.0008	0.0009	0.2576	0.9064	0.9326	0.0010	*
* 85	0.9336	0.0013	0.9332	0.0015	1.0008	0.0009	0.2571	0.9071	0.9326	0.0010	*
* 86	0.9334	0.0013	0.9330	0.0015	1.0008	0.0009	0.2567	0.9072	0.9324	0.0010	*
* 87	0.9333	0.0013	0.9328	0.0015	1.0007	0.0009	0.2563	0.9070	0.9323	0.0010	*
* 88	0.9332	0.0013	0.9329	0.0015	1.0009	0.0009	0.2575	0.9058	0.9322	0.0010	*
* 89	0.9330	0.0013	0.9327	0.0015	1.0007	0.0009	0.2572	0.9039	0.9321	0.0010	*
* 90	0.9329	0.0013	0.9324	0.0015	1.0005	0.0008	0.2558	0.9041	0.9320	0.0010	*
* 91	0.9329	0.0013	0.9326	0.0014	1.0006	0.0008	0.2577	0.9042	0.9321	0.0010	*
* 92	0.9329	0.0013	0.9324	0.0014	1.0005	0.0008	0.2572	0.9042	0.9320	0.0010	*
* 93	0.9330	0.0013	0.9324	0.0014	1.0006	0.0008	0.2576	0.9038	0.9321	0.0010	*
* 94	0.9330	0.0013	0.9326	0.0014	1.0006	0.0008	0.2580	0.9040	0.9322	0.0010	*
* 95	0.9330	0.0012	0.9324	0.0014	1.0004	0.0008	0.2567	0.9043	0.9322	0.0010	*
* 96	0.9329	0.0012	0.9324	0.0014	1.0004	0.0008	0.2568	0.9041	0.9322	0.0010	*
* 97	0.9327	0.0012	0.9324	0.0014	1.0004	0.0008	0.2582	0.9024	0.9321	0.0010	*
* 98	0.9328	0.0012	0.9324	0.0014	1.0004	0.0008	0.2586	0.9018	0.9322	0.0010	*
* 99	0.9326	0.0012	0.9324	0.0014	1.0004	0.0008	0.2599	0.9000	0.9320	0.0010	*
* 100	0.9329	0.0012	0.9324	0.0014	1.0004	0.0008	0.2585	0.9017	0.9322	0.0009	*
* 101	0.9329	0.0012	0.9324	0.0014	1.0004	0.0008	0.2583	0.9031	0.9321	0.0009	*
* 102	0.9328	0.0012	0.9321	0.0014	1.0004	0.0008	0.2571	0.9031	0.9319	0.0009	*
* 103	0.9328	0.0012	0.9319	0.0014	1.0002	0.0008	0.2560	0.9032	0.9319	0.0009	*
* 104	0.9327	0.0012	0.9319	0.0014	1.0002	0.0008	0.2566	0.9045	0.9319	0.0009	*
* 105	0.9326	0.0012	0.9317	0.0013	1.0001	0.0008	0.2560	0.9041	0.9319	0.0009	*
* 106	0.9324	0.0012	0.9316	0.0013	1.0002	0.0008	0.2571	0.9028	0.9316	0.0009	*
* 107	0.9325	0.0012	0.9316	0.0013	1.0002	0.0008	0.2577	0.9028	0.9317	0.0009	*
* 108	0.9323	0.0012	0.9315	0.0013	1.0002	0.0008	0.2579	0.9011	0.9316	0.0009	*
* 109	0.9322	0.0012	0.9313	0.0013	1.0001	0.0008	0.2585	0.8998	0.9315	0.0009	*
* 110	0.9321	0.0012	0.9312	0.0013	1.0000	0.0008	0.2591	0.8989	0.9315	0.0009	*
* 111	0.9320	0.0012	0.9311	0.0013	1.0000	0.0008	0.2591	0.8987	0.9314	0.0009	*
* 112	0.9319	0.0012	0.9308	0.0013	0.9998	0.0008	0.2581	0.8979	0.9313	0.0009	*
* 113	0.9318	0.0011	0.9305	0.0013	0.9997	0.0008	0.2580	0.8973	0.9311	0.0009	*
* 114	0.9317	0.0011	0.9303	0.0013	0.9996	0.0008	0.2576	0.8956	0.9310	0.0009	*
* 115	0.9316	0.0011	0.9303	0.0013	0.9996	0.0008	0.2583	0.8943	0.9310	0.0009	*
* 116	0.9315	0.0011	0.9302	0.0013	0.9996	0.0007	0.2585	0.8944	0.9310	0.0009	*
* 117	0.9315	0.0011	0.9301	0.0013	0.9995	0.0007	0.2575	0.8960	0.9309	0.0009	*
* 118	0.9314	0.0011	0.9302	0.0013	0.9995	0.0007	0.2583	0.8941	0.9309	0.0009	*
* 119	0.9314	0.0011	0.9302	0.0013	0.9996	0.0007	0.2595	0.8925	0.9309	0.0009	*
* 120	0.9315	0.0011	0.9303	0.0013	0.9997	0.0007	0.2603	0.8925	0.9309	0.0009	*
* 121	0.9315	0.0011	0.9305	0.0013	0.9998	0.0007	0.2613	0.8919	0.9310	0.0009	*
* 122	0.9316	0.0011	0.9306	0.0013	0.9999	0.0007	0.2622	0.8911	0.9310	0.0009	*
* 123	0.9315	0.0011	0.9306	0.0012	0.9999	0.0007	0.2626	0.8916	0.9310	0.0009	*
* 124	0.9315	0.0011	0.9307	0.0012	0.9999	0.0007	0.2621	0.8914	0.9310	0.0009	*
* 125	0.9316	0.0011	0.9307	0.0012	0.9999	0.0007	0.2615	0.8913	0.9311	0.0009	*
* 126	0.9316	0.0011	0.9308	0.0012	0.9999	0.0007	0.2611	0.8921	0.9311	0.0008	*
* 127	0.9315	0.0011	0.9307	0.0012	0.9999	0.0007	0.2618	0.8908	0.9310	0.0008	*
* 128	0.9314	0.0011	0.9307	0.0012	0.9999	0.0007	0.2619	0.8903	0.9310	0.0008	*
* 129	0.9314	0.0011	0.9308	0.0012	1.0000	0.0007	0.2626	0.8898	0.9310	0.0008	*
* 130	0.9314	0.0011	0.9309	0.0012	1.0000	0.0007	0.2640	0.8893	0.9310	0.0008	*
* 131	0.9314	0.0011	0.9310	0.0012	1.0000	0.0007	0.2639	0.8902	0.9311	0.0008	*
* 132	0.9313	0.0011	0.9310	0.0012	1.0000	0.0007	0.2639	0.8897	0.9311	0.0008	*
* 133	0.9312	0.0011	0.9309	0.0012	1.0000	0.0007	0.2637	0.8900	0.9310	0.0008	*
* 134	0.9313	0.0011	0.9308	0.0012	1.0000	0.0007	0.2632	0.8913	0.9309	0.0008	*
* 135	0.9313	0.0010	0.9308	0.0012	1.0000	0.0007	0.2634	0.8914	0.9309	0.0008	*
* 136	0.9315	0.0010	0.9308	0.0012	1.0000	0.0007	0.2627	0.8926	0.9310	0.0008	*
* 137	0.9317	0.0010	0.9311	0.0012	1.0000	0.0007	0.2636	0.8924	0.9312	0.0008	*
* 138	0.9318	0.0010	0.9312	0.0012	1.0001	0.0007	0.2649	0.8915	0.9312	0.0008	*
* 139	0.9319	0.0010	0.9311	0.0012	1.0000	0.0007	0.2639	0.8922	0.9313	0.0008	*
* 140	0.9318	0.0010	0.9311	0.0012	1.0000	0.0007	0.2646	0.8914	0.9313	0.0008	*
* 141	0.9317	0.0010	0.9311	0.0012	1.0000	0.0007	0.2653	0.8897	0.9313	0.0008	*
* 142	0.9317	0.0010	0.9311	0.0012	1.0000	0.0007	0.2660	0.8886	0.9313	0.0008	*
1*****											
STAGE	K(COLL)	STDV	INDIVIDUAL K(SCORE)	STDV	STAGE	K-EFFECTIVE A(SCORE)	STDV	ESTIMATORS ALPHA	BETA	K(THREE)	STDV
* -3	0.8951	0.0140	0.9146	0.0159	1.0005	0.0087	0.3553	0.8636	0.9072	0.0106	*
* -2	0.9166	0.0117	0.9278	0.0135	1.0003	0.0078	0.3260	0.8554	0.9239	0.0089	*
* -1	0.9376	0.0131	0.9669	0.0144	1.0226	0.0086	0.2679	0.9749	0.9370	0.0089	*
* 0	0.9135	0.0130	0.9359	0.0169	1.0096	0.0095	0.4350	0.9297	0.9172	0.0096	*

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

*	1	0.9195	0.0129	0.9050	0.0124	0.9973	0.0077	0.0841	0.9162	0.9087	0.0096	*
*	2	0.9373	0.0114	0.9497	0.0121	1.0049	0.0071	0.2484	0.9644	0.9419	0.0081	*
*	3	0.9419	0.0113	0.9311	0.0137	0.9979	0.0081	0.2730	0.8389	0.9358	0.0093	*
*	4	0.9406	0.0125	0.9333	0.0131	0.9933	0.0072	0.2029	0.8754	0.9406	0.0099	*
*	5	0.9097	0.0120	0.9065	0.0157	0.9931	0.0084	0.3116	0.9744	0.9142	0.0100	*
*	6	0.9386	0.0119	0.9402	0.0148	1.0032	0.0086	0.3587	0.8313	0.9370	0.0099	*
*	7	0.9290	0.0121	0.9276	0.0139	0.9960	0.0083	0.2208	0.9440	0.9317	0.0095	*
*	8	0.9271	0.0123	0.9395	0.0135	0.9998	0.0086	0.2760	0.7975	0.9362	0.0091	*
*	9	0.9102	0.0120	0.9310	0.0125	1.0014	0.0077	0.4306	0.4924	0.9214	0.0092	*
*	10	0.9168	0.0133	0.9049	0.0150	0.9950	0.0105	0.0862	0.9637	0.9108	0.0101	*
*	11	0.9376	0.0113	0.9333	0.0129	1.0018	0.0077	0.2868	0.8682	0.9329	0.0086	*
*	12	0.9334	0.0119	0.9122	0.0150	0.9948	0.0093	0.1896	0.9959	0.9214	0.0098	*
*	13	0.9161	0.0139	0.9245	0.0140	1.0079	0.0075	0.2063	0.9937	0.9149	0.0102	*
*	14	0.9207	0.0124	0.9584	0.0127	1.0239	0.0081	0.4102	0.6152	0.9282	0.0088	*
*	15	0.9181	0.0135	0.9149	0.0154	1.0003	0.0078	0.2993	0.8925	0.9156	0.0115	*
*	16	0.9488	0.0105	0.9357	0.0130	0.9992	0.0074	0.3042	0.9064	0.9403	0.0084	*
*	17	0.9397	0.0108	0.9396	0.0126	1.0040	0.0079	0.3529	0.6970	0.9369	0.0085	*
*	18	0.9261	0.0129	0.9312	0.0146	1.0049	0.0085	0.2770	0.8820	0.9255	0.0103	*
*	19	0.9484	0.0099	0.9517	0.0108	1.0016	0.0068	0.2643	0.7812	0.9495	0.0074	*
*	20	0.9295	0.0123	0.9330	0.0146	0.9973	0.0073	0.3696	0.8931	0.9342	0.0101	*
*	21	0.9335	0.0107	0.9290	0.0134	0.9961	0.0080	0.2522	0.9922	0.9340	0.0081	*
*	22	0.9440	0.0135	0.9399	0.0141	1.0039	0.0078	0.0907	1.1192	0.9359	0.0102	*
*	23	0.9477	0.0134	0.9225	0.0133	0.9894	0.0073	0.0439	1.0775	0.9350	0.0104	*
*	24	0.9342	0.0112	0.9528	0.0135	1.0049	0.0076	0.4250	0.7931	0.9410	0.0087	*
*	25	0.9410	0.0114	0.9344	0.0134	1.0040	0.0073	0.2731	0.9427	0.9325	0.0093	*
*	26	0.9221	0.0129	0.9326	0.0139	1.0031	0.0082	0.3336	0.7299	0.9268	0.0104	*
*	27	0.9222	0.0110	0.9335	0.0133	1.0104	0.0078	0.3901	0.7173	0.9217	0.0082	*
*	28	0.9335	0.0124	0.9159	0.0150	0.9825	0.0075	0.3440	0.7236	0.9346	0.0116	*
*	29	0.9258	0.0118	0.9297	0.0152	1.0003	0.0082	0.3601	0.9382	0.9280	0.0095	*
*	30	0.9423	0.0120	0.9212	0.0125	0.9925	0.0073	0.1544	0.8211	0.9307	0.0099	*
*	31	0.9483	0.0104	0.9557	0.0124	1.0144	0.0078	0.3563	0.8204	0.9412	0.0078	*
*	32	0.9296	0.0142	0.9404	0.0157	0.9976	0.0080	0.1896	1.1882	0.9413	0.0109	*
*	33	0.9338	0.0133	0.9392	0.0130	0.9987	0.0081	0.1250	0.9802	0.9398	0.0091	*
*	34	0.9374	0.0123	0.9251	0.0138	0.9984	0.0086	0.1225	1.0395	0.9282	0.0094	*
*	35	0.9300	0.0122	0.9397	0.0144	1.0165	0.0085	0.2715	0.9898	0.9207	0.0089	*
*	36	0.9388	0.0115	0.9287	0.0139	0.9939	0.0076	0.1855	1.1751	0.9378	0.0089	*
*	37	0.9552	0.0112	0.9497	0.0127	1.0125	0.0073	0.2650	0.9421	0.9394	0.0085	*
*	38	0.9140	0.0136	0.9137	0.0159	1.0013	0.0090	0.2889	0.8661	0.9127	0.0114	*
*	39	0.9345	0.0125	0.9339	0.0140	0.9944	0.0080	0.2600	0.8631	0.9389	0.0100	*
*	40	0.9164	0.0104	0.9199	0.0142	0.9971	0.0086	0.4411	0.7032	0.9204	0.0085	*
*	41	0.9526	0.0103	0.9599	0.0119	1.0159	0.0075	0.3193	0.9036	0.9432	0.0072	*
*	42	0.9415	0.0111	0.9433	0.0139	1.0001	0.0082	0.4005	0.7294	0.9425	0.0093	*
*	43	0.9216	0.0113	0.9127	0.0131	0.9975	0.0088	0.2044	0.8667	0.9167	0.0086	*
*	44	0.9477	0.0117	0.9487	0.0124	1.0088	0.0072	0.2652	0.8369	0.9411	0.0089	*
*	45	0.9445	0.0116	0.9374	0.0136	1.0010	0.0077	0.2367	0.9730	0.9381	0.0089	*
*	46	0.9095	0.0124	0.9296	0.0143	0.9967	0.0079	0.3229	0.9268	0.9262	0.0093	*
*	47	0.9237	0.0144	0.9377	0.0146	1.0105	0.0081	0.1930	1.0598	0.9239	0.0102	*
*	48	0.9161	0.0141	0.9220	0.0142	1.0011	0.0084	0.1812	0.9084	0.9199	0.0105	*
*	49	0.9393	0.0116	0.9402	0.0142	1.0117	0.0089	0.3101	0.8215	0.9303	0.0092	*
*	50	0.9483	0.0113	0.9462	0.0137	1.0004	0.0083	0.2685	0.8802	0.9464	0.0084	*
*	51	0.9358	0.0129	0.9446	0.0143	1.0082	0.0087	0.2670	0.8964	0.9348	0.0094	*
*	52	0.9449	0.0114	0.9438	0.0143	1.0031	0.0081	0.3121	0.9253	0.9414	0.0090	*
*	53	0.9304	0.0117	0.9238	0.0135	0.9927	0.0082	0.2366	0.8668	0.9316	0.0096	*
*	54	0.9184	0.0114	0.9110	0.0144	0.9934	0.0087	0.3229	0.7995	0.9187	0.0094	*
*	55	0.9354	0.0116	0.9227	0.0134	0.9889	0.0081	0.1331	1.0369	0.9359	0.0090	*
*	56	0.9199	0.0124	0.9086	0.0150	0.9902	0.0081	0.2520	0.9731	0.9210	0.0106	*
*	57	0.9409	0.0110	0.9491	0.0136	1.0022	0.0081	0.3756	0.7595	0.9443	0.0084	*
*	58	0.9604	0.0110	0.9504	0.0125	1.0055	0.0077	0.2011	0.9601	0.9472	0.0081	*
*	59	0.9239	0.0110	0.9149	0.0151	0.9883	0.0090	0.4070	0.7126	0.9269	0.0102	*
*	60	0.9529	0.0132	0.9523	0.0129	1.0079	0.0076	0.1793	0.8312	0.9458	0.0100	*
*	61	0.9295	0.0124	0.9329	0.0145	0.9973	0.0089	0.2198	0.9589	0.9347	0.0093	*
*	62	0.9310	0.0122	0.9256	0.0133	1.0024	0.0081	0.2517	0.7817	0.9251	0.0096	*
*	63	0.9349	0.0119	0.9291	0.0130	1.0030	0.0075	0.2553	0.9134	0.9278	0.0087	*
*	64	0.9205	0.0127	0.9553	0.0145	1.0231	0.0076	0.5181	0.7500	0.9199	0.0093	*
*	65	0.9313	0.0125	0.9155	0.0131	0.9893	0.0079	0.1116	1.0426	0.9284	0.0091	*
*	66	0.9395	0.0122	0.9276	0.0135	0.9968	0.0082	0.1073	1.0772	0.9324	0.0090	*
*	67	0.9483	0.0121	0.9458	0.0142	1.0087	0.0087	0.2338	0.9858	0.9378	0.0086	*
*	68	0.9431	0.0114	0.9478	0.0139	1.0027	0.0080	0.3315	0.8804	0.9439	0.0085	*
*	69	0.9268	0.0122	0.9272	0.0148	0.9916	0.0084	0.3204	0.8270	0.9340	0.0102	*
*	70	0.9480	0.0141	0.9246	0.0134	0.9835	0.0074	-0.0515	1.1876	0.9430	0.0105	*
*	71	0.9232	0.0151	0.9569	0.0167	1.0130	0.0094	0.2734	1.0548	0.9339	0.0103	*
*	72	0.9485	0.0138	0.9395	0.0138	0.9961	0.0082	0.1036	1.0984	0.9447	0.0095	*
*	73	0.9090	0.0134	0.9005	0.0145	0.9871	0.0088	0.2461	0.6694	0.9112	0.0115	*
*	74	0.9260	0.0118	0.9145	0.0135	0.9924	0.0082	0.2623	0.7693	0.9233	0.0096	*

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

*	75	0.9413	0.0122	0.9493	0.0137	1.0110	0.0080	0.3276	0.9196	0.9366	0.0079	*
*	76	0.9354	0.0115	0.9396	0.0139	1.0078	0.0084	0.2986	0.8857	0.9314	0.0088	*
*	77	0.9041	0.0117	0.9268	0.0151	1.0064	0.0086	0.4818	0.7575	0.9110	0.0085	*
*	78	0.9415	0.0115	0.9378	0.0135	1.0029	0.0083	0.2484	0.9104	0.9361	0.0090	*
*	79	0.9196	0.0118	0.9378	0.0136	1.0114	0.0083	0.3701	0.7806	0.9221	0.0088	*
*	80	0.9435	0.0113	0.9359	0.0125	0.9909	0.0073	0.2476	0.8885	0.9459	0.0087	*
*	81	0.9347	0.0107	0.9340	0.0139	1.0022	0.0081	0.3262	0.9097	0.9322	0.0087	*
*	82	0.9365	0.0144	0.9436	0.0154	0.9996	0.0076	0.2178	1.1815	0.9426	0.0105	*
*	83	0.9421	0.0129	0.9194	0.0126	0.9890	0.0076	0.0021	1.1353	0.9319	0.0092	*
*	84	0.9146	0.0115	0.9237	0.0149	1.0040	0.0082	0.4612	0.6578	0.9169	0.0094	*
*	85	0.9398	0.0112	0.9279	0.0127	0.9951	0.0071	0.2039	0.9628	0.9350	0.0094	*
*	86	0.9150	0.0128	0.9168	0.0140	1.0002	0.0081	0.2267	0.9265	0.9162	0.0098	*
*	87	0.9268	0.0150	0.9115	0.0148	0.9989	0.0075	0.1963	0.8652	0.9155	0.0122	*
*	88	0.9180	0.0120	0.9453	0.0132	1.0130	0.0078	0.3842	0.7888	0.9246	0.0086	*
*	89	0.9184	0.0124	0.9101	0.0125	0.9840	0.0071	0.2519	0.5945	0.9217	0.0104	*
*	90	0.9233	0.0127	0.9025	0.0134	0.9860	0.0075	0.1304	0.8784	0.9176	0.0108	*
*	91	0.9359	0.0119	0.9516	0.0148	1.0093	0.0080	0.4398	0.8921	0.9364	0.0090	*
*	92	0.9292	0.0121	0.9159	0.0132	0.9910	0.0077	0.2073	0.8861	0.9266	0.0096	*
*	93	0.9402	0.0115	0.9378	0.0128	1.0032	0.0073	0.2851	0.8707	0.9356	0.0090	*
*	94	0.9344	0.0115	0.9456	0.0140	1.0019	0.0084	0.2832	0.9231	0.9407	0.0087	*
*	95	0.9352	0.0127	0.9147	0.0138	0.9806	0.0079	0.1366	0.9134	0.9352	0.0107	*
*	96	0.9209	0.0134	0.9376	0.0141	1.0049	0.0084	0.2687	0.8925	0.9288	0.0092	*
*	97	0.9067	0.0135	0.9307	0.0160	1.0037	0.0092	0.4413	0.6620	0.9176	0.0106	*
*	98	0.9432	0.0123	0.9302	0.0145	0.9960	0.0079	0.2927	0.8608	0.9375	0.0105	*
*	99	0.9137	0.0107	0.9312	0.0138	1.0076	0.0088	0.4137	0.6576	0.9190	0.0077	*
*	100	0.9559	0.0113	0.9338	0.0126	0.9956	0.0069	0.1557	1.0458	0.9418	0.0091	*
*	101	0.9339	0.0118	0.9306	0.0134	1.0022	0.0074	0.2474	1.0442	0.9291	0.0089	*
*	102	0.9241	0.0118	0.9016	0.0130	0.9956	0.0086	0.1404	0.9052	0.9088	0.0090	*
*	103	0.9315	0.0134	0.9111	0.0136	0.9858	0.0079	0.1678	0.8515	0.9266	0.0102	*
*	104	0.9267	0.0123	0.9345	0.0154	0.9979	0.0086	0.3170	1.0249	0.9342	0.0092	*
*	105	0.9236	0.0126	0.9151	0.0138	0.9893	0.0083	0.1919	0.8749	0.9261	0.0102	*
*	106	0.8991	0.0132	0.9148	0.0149	1.0113	0.0084	0.4125	0.7450	0.8999	0.0102	*
*	107	0.9428	0.0120	0.9316	0.0147	0.9993	0.0083	0.3054	0.9145	0.9357	0.0097	*
*	108	0.9094	0.0144	0.9176	0.0151	0.9984	0.0093	0.2898	0.6919	0.9163	0.0114	*
*	109	0.9160	0.0120	0.9115	0.0142	0.9881	0.0080	0.3568	0.6876	0.9213	0.0105	*
*	110	0.9255	0.0120	0.9211	0.0146	0.9904	0.0080	0.3496	0.7460	0.9298	0.0107	*
*	111	0.9192	0.0119	0.9181	0.0138	0.9975	0.0079	0.2709	0.8695	0.9206	0.0095	*
*	112	0.9267	0.0119	0.8961	0.0119	0.9845	0.0073	0.1318	0.7111	0.9111	0.0099	*
*	113	0.9184	0.0119	0.8985	0.0133	0.9914	0.0080	0.2282	0.8004	0.9099	0.0100	*
*	114	0.9122	0.0130	0.9051	0.0130	0.9876	0.0083	0.2296	0.6337	0.9146	0.0101	*
*	115	0.9277	0.0115	0.9261	0.0134	0.9941	0.0075	0.3419	0.7219	0.9309	0.0096	*
*	116	0.9214	0.0118	0.9231	0.0137	0.9949	0.0075	0.2997	0.8945	0.9272	0.0095	*
*	117	0.9255	0.0133	0.9084	0.0145	0.9920	0.0083	0.1307	1.0806	0.9193	0.0104	*
*	118	0.9252	0.0117	0.9424	0.0122	1.0027	0.0071	0.3654	0.6279	0.9344	0.0087	*
*	119	0.9307	0.0110	0.9285	0.0137	1.0047	0.0081	0.3850	0.7430	0.9259	0.0089	*
*	120	0.9355	0.0132	0.9535	0.0150	1.0122	0.0085	0.3537	0.8683	0.9366	0.0098	*
*	121	0.9358	0.0112	0.9449	0.0136	1.0093	0.0075	0.3685	0.8108	0.9340	0.0087	*
*	122	0.9386	0.0116	0.9469	0.0136	1.0124	0.0081	0.3579	0.8001	0.9340	0.0087	*
*	123	0.9213	0.0129	0.9337	0.0153	1.0036	0.0087	0.3232	0.9504	0.9262	0.0097	*
*	124	0.9313	0.0132	0.9391	0.0138	1.0028	0.0085	0.2088	0.8743	0.9350	0.0098	*
*	125	0.9413	0.0131	0.9357	0.0137	0.9972	0.0081	0.1981	0.8731	0.9392	0.0099	*
*	126	0.9356	0.0113	0.9377	0.0124	0.9980	0.0074	0.2058	0.9977	0.9392	0.0081	*
*	127	0.9185	0.0112	0.9212	0.0135	1.0087	0.0084	0.3678	0.7453	0.9137	0.0090	*
*	128	0.9142	0.0131	0.9229	0.0139	0.9912	0.0079	0.2869	0.7784	0.9272	0.0100	*
*	129	0.9295	0.0118	0.9479	0.0141	1.0103	0.0086	0.3769	0.8090	0.9326	0.0089	*
*	130	0.9336	0.0104	0.9432	0.0139	1.0059	0.0080	0.4195	0.8109	0.9344	0.0079	*
*	131	0.9355	0.0123	0.9379	0.0137	1.0003	0.0072	0.2500	1.0264	0.9371	0.0095	*
*	132	0.9218	0.0126	0.9409	0.0123	1.0009	0.0076	0.2611	0.8039	0.9352	0.0084	*
*	133	0.9184	0.0125	0.9196	0.0147	1.0002	0.0091	0.2331	0.9375	0.9191	0.0093	*
*	134	0.9341	0.0120	0.9178	0.0140	0.9991	0.0074	0.1709	1.1236	0.9215	0.0097	*
*	135	0.9394	0.0105	0.9188	0.0138	0.9926	0.0080	0.2821	0.8896	0.9312	0.0092	*
*	136	0.9556	0.0107	0.9384	0.0133	1.0045	0.0081	0.1978	1.0134	0.9372	0.0085	*
*	137	0.9500	0.0113	0.9603	0.0133	1.0033	0.0074	0.3475	0.8754	0.9538	0.0087	*
*	138	0.9476	0.0107	0.9420	0.0138	1.0096	0.0077	0.4184	0.7942	0.9367	0.0092	*
*	139	0.9485	0.0120	0.9310	0.0130	0.9931	0.0074	0.1376	0.9936	0.9402	0.0097	*
*	140	0.9101	0.0132	0.9243	0.0150	0.9927	0.0085	0.3846	0.7448	0.9243	0.0106	*
*	141	0.9222	0.0119	0.9245	0.0134	0.9932	0.0076	0.3820	0.6121	0.9278	0.0099	*
*	142	0.9271	0.0111	0.9417	0.0134	1.0027	0.0084	0.3761	0.6994	0.9343	0.0084	*

***** NEUTRON BALANCE SUMMARY *****												
***** INDIVIDUAL STAGE COUNTS ***** CUMULATIVE COUNTS *****												
***** STAGE SAMPLE SCATTER CAPTURE FISSION CHILDREN N,2N ESCAPE *****												

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)																
*	-3	7709	303968	4900	2812	6900	3	0	*							*
*	-2	9933	388745	6234	3708	9105	8	0	*							*
*	-1	9450	371158	5851	3609	8860	10	0	*							*
*	0	9614	378133	6041	3580	8783	7	0	*							*
*	1	9536	370098	5970	3574	8769	8	0	*	9536	370098	5970	3574	8769	8	0
*	2	11231	423892	6952	4288	10527	9	0	*	20767	793990	12922	7862	19295	17	0
*	3	10377	411650	6410	3978	9774	11	0	*	31144	1205640	19332	11840	29069	28	0
*	4	10321	396951	6381	3952	9708	12	0	*	41465	1602591	25713	15792	38777	40	0
*	5	9184	358888	5789	3402	8354	7	0	*	50649	1961479	31502	19194	47131	47	0
*	6	10490	404280	6490	4008	9846	8	0	*	61139	2365759	37992	23202	56977	55	0
*	7	10082	395128	6281	3811	9366	10	0	*	71221	2760887	44273	27013	66343	65	0
*	8	9689	377504	6046	3653	8982	10	0	*	80910	3138391	50319	30666	75326	75	0
*	9	9379	374832	5915	3475	8537	11	0	*	90289	3513223	56234	34141	83862	86	0
*	10	9104	356214	5713	3399	8347	8	0	*	99393	3869437	61947	37540	92209	94	0
*	11	10159	397342	6299	3874	9525	14	0	*	109552	4266779	68246	41414	101734	108	0
*	12	8997	382650	6143	3761	9238	7	0	*	119449	4649429	74389	45175	110972	115	0
*	13	9091	357957	5714	3389	8328	12	0	*	128540	5007386	80103	48564	119300	127	0
*	14	9295	369206	5816	3484	8558	5	0	*	137835	5376592	85919	52048	127857	132	0
*	15	9193	362142	5766	3435	8440	8	0	*	147028	5738734	91685	55483	136297	140	0
*	16	11400	442837	7008	4402	10816	10	0	*	158428	6181571	98693	59885	147113	150	0
*	17	11079	439274	6847	4238	10411	6	0	*	169507	6620845	105540	64123	157524	156	0
*	18	9655	387133	6016	3643	8942	4	0	*	179162	7007978	111556	67766	166466	160	0
*	19	12256	479280	7532	4732	11624	8	0	*	191418	7487258	119088	72498	178089	168	0
*	20	10435	413150	6501	3944	9699	10	0	*	201853	7900408	125589	76442	187789	178	0
*	21	10596	408462	6580	4028	9891	12	0	*	212449	8308870	132169	80470	197680	190	0
*	22	11059	429889	6817	4251	10440	9	0	*	223508	8738759	138986	84721	208120	199	0
*	23	10692	410293	6581	4125	10133	14	0	*	234200	9149052	145567	88846	218253	213	0
*	24	10046	387383	6230	3820	9384	4	0	*	244246	9536435	151797	92666	227637	217	0
*	25	10651	414808	6578	4081	10023	8	0	*	254897	9951243	158375	96747	237660	225	0
*	26	9471	375338	5926	3556	8733	11	0	*	264368	10326581	164301	100303	246394	236	0
*	27	9755	385220	6096	3665	8996	6	0	*	274123	10711801	170397	103968	255390	242	0
*	28	9954	381461	6187	3781	9292	14	0	*	284077	11093262	176584	107749	264682	256	0
*	29	9731	374928	6072	3670	9009	11	0	*	293808	11468190	182656	111419	273691	267	0
*	30	11135	441523	6874	4270	10493	9	0	*	304943	11909713	189530	115689	284184	276	0
*	31	10997	424271	6759	4248	10428	10	0	*	315940	12333984	196289	119937	294612	286	0
*	32	9548	366071	5941	3610	8876	3	0	*	325488	12700055	202230	123547	303487	289	0
*	33	10246	399032	6361	3893	9568	8	0	*	335734	13099087	208591	127440	313055	297	0
*	34	9957	385519	6159	3801	9334	3	0	*	345691	13484606	214750	131241	322389	300	0
*	35	9793	386211	6088	3711	9107	6	0	*	355484	13870817	220838	134952	331496	306	0
*	36	10613	409387	6560	4060	9964	7	0	*	366097	14280204	227398	139012	341460	313	0
*	37	10977	433629	6717	4269	10486	9	0	*	377074	14713833	234115	143281	351946	322	0
*	38	8678	342907	5462	3224	7932	8	0	*	385752	15056740	239577	146505	359878	330	0
*	39	10165	392203	6301	3866	9499	2	0	*	395917	15448943	245878	150371	369377	332	0
*	40	9784	379996	6146	3649	8966	11	0	*	405701	15828939	252024	154020	378343	343	0
*	41	11756	454332	7202	4564	11198	10	0	*	417457	16283271	259226	158584	389541	353	0
*	42	10567	411971	6523	4052	9949	8	0	*	428024	16695242	265749	162636	399490	361	0
*	43	9802	382644	6130	3679	9034	7	0	*	437826	17077886	271879	166315	408524	368	0
*	44	11285	447123	6933	4359	10695	7	0	*	449111	17525009	278812	170674	419218	375	0
*	45	10776	420644	6637	4147	10178	8	0	*	459887	17945653	285449	174821	429396	383	0
*	46	9187	355157	5795	3400	8355	8	0	*	469074	18300810	291244	178221	437752	391	0
*	47	9320	369224	5820	3507	8609	7	0	*	478394	18670034	297064	181728	446361	398	0
*	48	9125	358547	5732	3404	8360	11	0	*	487519	19028581	302796	185132	454720	409	0
*	49	9833	387945	6084	3757	9236	8	0	*	497352	19416526	308880	188889	463956	417	0
*	50	10863	424013	6680	4192	10302	9	0	*	508215	19840539	315560	193081	474258	426	0
*	51	9265	365015	5743	3528	8670	6	0	*	517480	20205554	321303	196609	482928	432	0
*	52	10751	422652	6629	4130	10159	8	0	*	528231	20628206	327932	200739	493087	440	0
*	53	10044	383143	6247	3803	9345	6	0	*	538275	21011349	334179	204542	502432	446	0
*	54	9340	364276	5853	3493	8578	6	0	*	547615	21375625	340032	208035	511010	452	0
*	55	10180	387203	6310	3878	9522	8	0	*	557795	21762828	346342	211913	520532	460	0
*	56	9401	360928	5889	3519	8648	7	0	*	567196	22123756	352231	215432	529180	467	0
*	57	11368	445909	7026	4350	10697	8	0	*	578564	22569665	359257	219782	539876	475	0
*	58	11943	467670	7282	4669	11469	8	0	*	590507	23037335	366539	224451	551346	483	0
*	59	10103	387518	6305	3802	9335	4	0	*	600610	23424853	372844	228253	560680	487	0
*	60	10551	413239	6465	4094	10054	8	0	*	611161	23838092	379309	232347	570735	495	0
*	61	9750	377179	6063	3691	9062	4	0	*	620911	24215271	385372	236038	579797	499	0
*	62	9538	373044	5928	3614	8880	4	0	*	630449	24588315	391300	239652	588677	503	0
*	63	10320	401207	6396	3930	9648	6	0	*	640769	24989522	397696	243582	598325	509	0
*	64	9653	381037	6045	3616	8886	8	0	*	650422	25370559	403741	247198	607210	517	0
*	65	10529	403921	6550	3992	9805	13	0	*	660951	25774480	410291	251190	617016	530	0
*	66	10152	389189	6284	3881	9538	12	0	*	671103	26163669	416575	255071	626554	542	0
*	67	10415	415196	6401	4017	9876	3	0	*	681518	26578865	422976	259088	636430	545	0
*	68	10299	395590	6355	3954	9713	10	0	*	691817	26974455	429331	263042	646144	555	0
*	69	9721	376525	6063	3668	9009	10	0	*	701538	27350980	435394	266710	655153	565	0
*	70	9747	373318	5996	3762	9240	11	0	*	711285	27724298	441390	270472	664393	576	0

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity

Condition (continued)																	
*	71	9406	364063	5878	3534	8683	6	0	*	720691	28088361	447268	274006	673077	582	0	*
*	72	10157	390441	6241	3923	9634	7	0	*	730848	28478802	453509	277929	682710	589	0	*
*	73	8831	342957	5571	3271	8028	11	0	*	739679	28821759	459080	281200	690738	600	0	*
*	74	9587	365668	5976	3618	8878	7	0	*	749266	29187427	465056	284818	699616	607	0	*
*	75	10106	393760	6241	3874	9512	9	0	*	759372	29581187	471297	288692	709128	616	0	*
*	76	10476	407282	6505	3984	9799	13	0	*	769848	29988469	477802	292676	718927	629	0	*
*	77	8945	349689	5657	3292	8088	4	0	*	778793	30338158	483459	295968	727015	633	0	*
*	78	10529	409877	6503	4034	9913	8	0	*	789322	30748035	489962	300002	736928	641	0	*
*	79	9755	381335	6116	3651	8971	12	0	*	799077	31129370	496078	303653	745899	653	0	*
*	80	10691	402590	6590	4106	10087	5	0	*	809768	31531960	502668	307759	755986	658	0	*
*	81	10775	421272	6680	4097	10072	2	0	*	820543	31953232	509348	311856	766058	660	0	*
*	82	9152	353106	5666	3489	8570	3	0	*	829695	32306338	515014	315345	774628	663	0	*
*	83	10506	405308	6492	4028	9897	14	0	*	840201	32711646	521506	319373	784525	677	0	*
*	84	9627	376797	6047	3588	8804	8	0	*	849828	33088443	527553	322961	793330	685	0	*
*	85	10743	418817	6642	4110	10096	9	0	*	860571	33507260	534195	327071	803426	694	0	*
*	86	9061	353704	5692	3372	8291	3	0	*	869632	33860964	539887	330443	811717	697	0	*
*	87	8808	344454	5495	3325	8164	12	0	*	878440	34205418	545382	333768	819881	709	0	*
*	88	9702	386274	6084	3628	8906	10	0	*	888142	34591692	551466	337396	828787	719	0	*
*	89	9882	383794	6206	3691	9076	15	0	*	898024	34975486	557672	341087	837863	734	0	*
*	90	9375	368735	5859	3522	8656	6	0	*	907399	35344221	563531	344609	846519	740	0	*
*	91	9937	381849	6159	3789	9300	11	0	*	917336	35726070	569690	348398	855818	751	0	*
*	92	9655	375033	6013	3651	8972	9	0	*	926991	36101103	575703	352049	864790	760	0	*
*	93	10470	414806	6471	4010	9844	10	0	*	937461	36515909	582174	356059	874634	770	0	*
*	94	10145	395351	6292	3863	9479	10	0	*	947606	36911260	588466	359922	884113	780	0	*
*	95	10141	396303	6290	3858	9484	6	0	*	957747	37307563	594756	363780	893597	786	0	*
*	96	9144	354763	5726	3428	8420	10	0	*	966891	37662326	600482	367208	902017	796	0	*
*	97	8700	339068	5497	3213	7888	10	0	*	975591	38001394	605979	370421	909905	806	0	*
*	98	10503	419408	6479	4034	9907	10	0	*	986094	38420802	612458	374455	919812	816	0	*
*	99	9490	373098	5968	3532	8671	10	0	*	995584	38793900	618426	377987	928482	826	0	*
*	100	11597	457272	7099	4513	11085	15	0	*	1007181	39251172	625525	382500	939567	841	0	*
*	101	10223	401422	6348	3886	9548	11	0	*	1017404	39652594	631873	386386	949115	852	0	*
*	102	10794	428171	6739	4062	9974	7	0	*	1028198	40080765	638612	390448	959089	859	0	*
*	103	9424	366423	5856	3575	8778	7	0	*	1037622	40447188	644468	394023	967868	866	0	*
*	104	9613	379999	6000	3625	8909	12	0	*	1047235	40827187	650468	397648	976776	878	0	*
*	105	10085	394389	6299	3794	9314	8	0	*	1057320	41221576	656767	401442	986091	886	0	*
*	106	8326	331291	5288	3046	7486	8	0	*	1065646	41552867	662055	404488	993577	894	0	*
*	107	10726	420214	6620	4121	10112	14	0	*	1076372	41973081	668675	408609	1003689	908	0	*
*	108	8329	334121	5251	3083	7574	5	0	*	1084701	42307202	673926	411692	1011263	913	0	*
*	109	9668	376839	6074	3604	8855	10	0	*	1094369	42684041	680000	415296	1020119	923	0	*
*	110	10170	391351	6344	3832	9412	6	0	*	1104539	43075392	686344	419128	1029531	929	0	*
*	111	9621	377706	6028	3598	8844	5	0	*	1114160	43453098	692372	422726	1038375	934	0	*
*	112	9806	385311	6112	3700	9087	6	0	*	1123966	43838409	698484	426426	1047462	940	0	*
*	113	9480	378729	5944	3547	8706	11	0	*	1133446	44216238	704428	429973	1056168	951	0	*
*	114	9165	357364	5771	3400	8360	6	0	*	1142611	44573602	710199	433373	1064528	957	0	*
*	115	10419	413664	6491	3939	9666	11	0	*	1153030	44987266	716690	437312	1074193	968	0	*
*	116	9795	375112	6126	3676	9025	7	0	*	1162825	45362378	722816	440988	1083218	975	0	*
*	117	9341	360925	5831	3519	8645	9	0	*	1172166	45723303	728647	444507	1091864	984	0	*
*	118	10422	398939	6501	3929	9643	8	0	*	1182588	46122242	735148	448436	1101506	992	0	*
*	119	10291	405633	6398	3902	9578	9	0	*	1192879	46527875	741546	452338	1111084	1001	0	*
*	120	9525	371059	5900	3630	8911	5	0	*	1202404	46898934	747446	455968	1119995	1006	0	*
*	121	10765	415781	6673	4101	10074	9	0	*	1213169	47314715	754119	460069	1130068	1015	0	*
*	122	10070	393674	6224	3851	9452	5	0	*	1223239	47708389	760343	463920	1139520	1020	0	*
*	123	9285	357912	5816	3484	8555	14	0	*	1232524	48066301	766159	467404	1148075	1034	0	*
*	124	9810	378858	6099	3718	9136	7	0	*	1242334	48445159	772258	471122	1157211	1041	0	*
*	125	9798	382815	6052	3754	9222	7	0	*	1252132	48827974	778310	474876	1166433	1048	0	*
*	126	10566	402549	6547	4028	9886	9	0	*	1262698	49230523	784857	478904	1176319	1057	0	*
*	127	9934	388706	6228	3717	9125	11	0	*	1272632	49619229	791085	482621	1185444	1068	0	*
*	128	9068	354462	5701	3373	8290	6	0	*	1281700	49973691	796786	485994	1193733	1074	0	*
*	129	9682	381840	6028	3662	9000	8	0	*	1291382	50355531	802814	489656	1202733	1082	0	*
*	130	10651	414819	6614	4049	9944	12	0	*	1302033	50770350	809428	493705	1212677	1094	0	*
*	131	10541	405168	6531	4018	9861	8	0	*	1312574	51175518	815959	497723	1222538	1102	0	*
*	132	9645	380049	6028	3623	8891	5	0	*	1322219	51555567	821987	501346	1231429	1107	0	*
*	133	9425	370050	5905	3527	8656	7	0	*	1331644	51925617	827892	504873	1240084	1114	0	*
*	134	10136	388817	6293	3853	9468	10	0	*	1341780	52314434	834185	508726	1249553	1124	0	*
*	135	10866	422519	6718	4154	10208	6	0	*	1352646	52736953	840903	512880	1259760	1130	0	*
*	136	11492	450746	7026	4475	10982	9	0	*	1364138	53187699	847929	517355	1270742	1139	0	*
*	137	11319	433364	6960	4375	10753	16	0	*	1375457	53621063	854889	521730	1281495	1155	0	*
*	138	11593	443518	7123	4475	10985	5	0	*	1387050	54064581	862012	526205	1292481	1160	0	*
*	139	10690	422102	6568	4127	10139	5	0	*	1397740	54486683	868580	530332	1302620	1165	0	*
*	140	9075	349495	5723	3360	8259	8	0	*	1406815	54836178	874303	533692	1310879	1173	0	*
*	141	9566	368040	5986	3591	8822	11	0	*	1416381	55204218	880289	537283	1319701	1184	0	*
*	142	9911	386935	6173	3745	9188	7	0	*	1426292	55591153	886462	541028	1328889	1191	0	*

1PLOT OF CUMULATIVE K(THREE) ESTIMATOR AGAINST STAGE NUMBER

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

	0.900	0.905	0.910	0.915	0.920	0.925	0.930	0.935	0.940	0.945	0.950	0.955	0.960
1	I	I	I	I	I	I	I	I	I	I	I	I	I
2	I	I	I	I	I	I	I	I	I	I	I	I	I
3	I	I	I	I	I	I	I	I	I	I	I	I	I
4	I	I	I	I	I	I	I	I	I	I	I	I	I
5	I	I	I	I	I	I	I	I	I	I	I	I	I
6	I	I	I	I	I	I	I	I	I	I	I	I	I
7	I	I	I	I	I	I	I	I	I	I	I	I	I
8	I	I	I	I	I	I	I	I	I	I	I	I	I
9	I	I	I	I	I	I	I	I	I	I	I	I	I
10	I	I	I	I	I	I	I	I	I	I	I	I	I
11	I	I	I	I	I	I	I	I	I	I	I	I	I
12	I	I	I	I	I	I	I	I	I	I	I	I	I
13	I	I	I	I	I	I	I	I	I	I	I	I	I
14	I	I	I	I	I	I	I	I	I	I	I	I	I
15	I	I	I	I	I	I	I	I	I	I	I	I	I
16	I	I	I	I	I	I	I	I	I	I	I	I	I
17	I	I	I	I	I	I	I	I	I	I	I	I	I
18	I	I	I	I	I	I	I	I	I	I	I	I	I
19	I	I	I	I	I	I	I	I	I	I	I	I	I
20	I	I	I	I	I	I	I	I	I	I	I	I	I
21	I	I	I	I	I	I	I	I	I	I	I	I	I
22	I	I	I	I	I	I	I	I	I	I	I	I	I
23	I	I	I	I	I	I	I	I	I	I	I	I	I
24	I	I	I	I	I	I	I	I	I	I	I	I	I
25	I	I	I	I	I	I	I	I	I	I	I	I	I
26	I	I	I	I	I	I	I	I	I	I	I	I	I
27	I	I	I	I	I	I	I	I	I	I	I	I	I
28	I	I	I	I	I	I	I	I	I	I	I	I	I
29	I	I	I	I	I	I	I	I	I	I	I	I	I
30	I	I	I	I	I	I	I	I	I	I	I	I	I
31	I	I	I	I	I	I	I	I	I	I	I	I	I
32	I	I	I	I	I	I	I	I	I	I	I	I	I
33	I	I	I	I	I	I	I	I	I	I	I	I	I
34	I	I	I	I	I	I	I	I	I	I	I	I	I
35	I	I	I	I	I	I	I	I	I	I	I	I	I
36	I	I	I	I	I	I	I	I	I	I	I	I	I
37	I	I	I	I	I	I	I	I	I	I	I	I	I
38	I	I	I	I	I	I	I	I	I	I	I	I	I
39	I	I	I	I	I	I	I	I	I	I	I	I	I
40	I	I	I	I	I	I	I	I	I	I	I	I	I
41	I	I	I	I	I	I	I	I	I	I	I	I	I
42	I	I	I	I	I	I	I	I	I	I	I	I	I
43	I	I	I	I	I	I	I	I	I	I	I	I	I
44	I	I	I	I	I	I	I	I	I	I	I	I	I
45	I	I	I	I	I	I	I	I	I	I	I	I	I
46	I	I	I	I	I	I	I	I	I	I	I	I	I
47	I	I	I	I	I	I	I	I	I	I	I	I	I
48	I	I	I	I	I	I	I	I	I	I	I	I	I
49	I	I	I	I	I	I	I	I	I	I	I	I	I
50	I	I	I	I	I	I	I	I	I	I	I	I	I
51	I	I	I	I	I	I	I	I	I	I	I	I	I
52	I	I	I	I	I	I	I	I	I	I	I	I	I
53	I	I	I	I	I	I	I	I	I	I	I	I	I
54	I	I	I	I	I	I	I	I	I	I	I	I	I
55	I	I	I	I	I	I	I	I	I	I	I	I	I
56	I	I	I	I	I	I	I	I	I	I	I	I	I
57	I	I	I	I	I	I	I	I	I	I	I	I	I
58	I	I	I	I	I	I	I	I	I	I	I	I	I
59	I	I	I	I	I	I	I	I	I	I	I	I	I
60	I	I	I	I	I	I	I	I	I	I	I	I	I
61	I	I	I	I	I	I	I	I	I	I	I	I	I
62	I	I	I	I	I	I	I	I	I	I	I	I	I
63	I	I	I	I	I	I	I	I	I	I	I	I	I
64	I	I	I	I	I	I	I	I	I	I	I	I	I
65	I	I	I	I	I	I	I	I	I	I	I	I	I
66	I	I	I	I	I	I	I	I	I	I	I	I	I
67	I	I	I	I	I	I	I	I	I	I	I	I	I
68	I	I	I	I	I	I	I	I	I	I	I	I	I
69	I	I	I	I	I	I	I	I	I	I	I	I	I
70	I	I	I	I	I	I	I	I	I	I	I	I	I
71	I	I	I	I	I	I	I	I	I	I	I	I	I

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity Condition (continued)

[illegible]

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** ..... **
** FINAL VALUE OF K(THREE) = 0.9313 ( STDV = 0.0008 ) **
** ..... **
** ..... **
** ..... **
** ..... **
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6.7-263

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

MATERIAL 8	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	78	1	0	0	0	0	79
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	3.76073E-02	9.91264E-08	0.00000E+00	4.26081E+00	0.00000E+00	0.00000E+00	4.11879E-02
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	2.21942E-07	2.81823E-08	0.00000E+00	3.62755E+00	0.00000E+00	0.00000E+00	2.19275E-07
MATERIAL 9	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	2178	74	0	72	0	0	2324
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	2.99537E-01	1.04150E-01	0.00000E+00	2.99848E+00	9.20106E+00	0.00000E+00	3.76362E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	5.48850E-04	1.79516E-07	0.00000E+00	2.47153E+00	9.20106E+00	0.00000E+00	5.50012E-04
MATERIAL 10	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	804	2438	0	11	0	0	3253
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	5.28858E-01	3.13167E-03	0.00000E+00	3.61397E+00	0.00000E+00	0.00000E+00	1.44993E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	1.35957E-02	2.73188E-07	0.00000E+00	3.17318E+00	0.00000E+00	0.00000E+00	4.17480E-06
MATERIAL 11	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	236	2	0	9	0	0	247
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	6.70501E-01	5.22666E-01	0.00000E+00	3.37984E+00	0.00000E+00	0.00000E+00	7.64902E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	5.68548E-02	8.69063E-07	0.00000E+00	2.97044E+00	0.00000E+00	0.00000E+00	6.10537E-02
MATERIAL 12	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	4414	348	0	97	0	0	4859
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	1.99235E-01	2.66845E-02	0.00000E+00	2.96924E+00	1.28827E+01	0.00000E+00	2.42245E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	3.51658E-05	6.17651E-08	0.00000E+00	2.43490E+00	1.28708E+01	0.00000E+00	2.78969E-05
MATERIAL 13	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	10	1	0	0	0	0	11
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	1.48663E-01	2.90288E-07	0.00000E+00	2.78299E+00	0.00000E+00	0.00000E+00	1.72410E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	6.69739E-06	4.38876E-08	0.00000E+00	2.03083E+00	0.00000E+00	0.00000E+00	5.06549E-06
MATERIAL 14	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	1178	37	0	33	0	0	1248
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	5.24631E-01	8.37436E-02	0.00000E+00	3.42078E+00	0.00000E+00	0.00000E+00	5.88168E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	2.18188E-03	5.05591E-08	0.00000E+00	3.02060E+00	0.00000E+00	0.00000E+00	1.92676E-03
MATERIAL 15	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	535	46	0	9	0	0	590
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	1.60907E-01	1.80555E-02	0.00000E+00	2.96621E+00	0.00000E+00	0.00000E+00	1.91034E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	1.85183E-05	6.07273E-08	0.00000E+00	2.33399E+00	0.00000E+00	0.00000E+00	1.41039E-05

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

MATERIAL 16	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	489	19	0	10	0	0	518
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	1.99776E-01	5.03783E-02	0.00000E+00	2.57752E+00	8.44186E+00	0.00000E+00	2.39550E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	2.61709E-04	1.68898E-07	0.00000E+00	1.98260E+00	8.44186E+00	0.00000E+00	2.35646E-04
MATERIAL 17	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	605	1	0	9	0	0	615
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	3.19062E-01	3.37821E-02	0.00000E+00	3.16794E+00	9.51841E+00	0.00000E+00	3.61278E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	8.12752E-04	2.93113E-07	0.00000E+00	2.67219E+00	9.44834E+00	0.00000E+00	9.01029E-04
MATERIAL 18	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	627	24	0	0	0	0	651
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	1.03265E-01	1.20080E-02	0.00000E+00	3.68962E+00	0.00000E+00	0.00000E+00	1.01454E-01
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	2.68269E-05	7.99855E-08	0.00000E+00	2.95077E+00	0.00000E+00	0.00000E+00	2.18204E-05
MATERIAL 19	ELASTIC	CAPTURE	FISSION	(N,N*)	(N,2N)	(N,3N)	TOTAL
TOTAL NO. OF EVENTS	733	3	0	0	0	0	736
MEAN ENERGY OF NEUTRONS CAUSING EVENTS (MEV)	7.83762E-02	8.00542E-02	0.00000E+00	7.67502E+00	0.00000E+00	0.00000E+00	7.85274E-02
MEAN LOG(ENERGY) OF NEUTRONS CAUSING EVENTS (MEV)	3.08455E-06	5.90435E-08	0.00000E+00	7.67253E+00	0.00000E+00	0.00000E+00	3.03881E-06

LCASE CATEGORISATION
=====

THERE ARE 972 CATEGORIES SPLIT AS FOLLOWS :

A=1:URANIUM SYSTEMS..... 1 - 324
A=2:PLUTONIUM SYSTEMS..... 325 - 648
A=3:OTHER SYSTEMS..... 649 - 972

WITHIN EACH 324 NUMBER SEGMENT :

B=1:LOW NON-FUEL ABSORPTION..... 1 - 108
B=2:MEDIUM NON-FUEL ABSORPTION... 109 - 216
B=3:HIG HIGH NON-FUEL ABSORPTION..... 217 - 324

WITHIN EACH 108 NUMBER SEGMENT :

C=1:LOW ASSEMBLIES..... 1 - 36
C=2:MEDIUM LEAKAGE SYSTEMS..... 37 - 72
C=3:HIG HIGH LEAKAGE SYSTEMS..... 73 - 108

WITHIN EACH 36 NUMBER SEGMENT :

D=1:LOW RESONANCE ABSORPTION..... 1 - 12
D=2:MEDIUM RESONANCE ABSORPTION... 13 - 24
D=3:HIG HIGH RESONANCE ABSORPTION..... 25 - 36

WITHIN EACH 12 NUMBER SEGMENT :

E=1:LOW FAST FISSION..... 1 - 4
E=2:MEDIUM FAST FISSION..... 5 - 8
E=3:HIG HIGH FAST FISSION..... 9 - 12

WITHIN EACH 4 NUMBER SEGMENT :

F=1:NO HYDROGEN..... 1
F=2:LOW HYDROGEN CONTENT..... 2
F=3:MEDIUM HYDROGEN CONTENT..... 3
F=4:HIG HIGH HYDROGEN CONTENT..... 4

TYPE OF SYSTEM	A = 1	FRACTION OF FISSIONS IN URANIUM AND PLUTONIUM = 1.0000 0.0000
NON-FUEL ABSORPTION	B = 2	FRACTION OF TOTAL ABSORPTIONS IN FUEL = 0.5805
LEAKAGE	C = 1	FRACTION OF NEUTRONS LEAKING = 0.0000
RESONANCE ABSORPTION	D = 3	FRACTION OF ABSORPTIONS IN RESONANCE PARTITION = 0.2563
FAST FISSION	E = 1	MEASURE OF FAST FISSION = 0.0284
FUEL HYDROGEN CONTENT	F = 1	MEASURE OF HYDROGEN CONTENT = 0.0000

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

* THIS CASE FALLS INTO CATEGORY NUMBER 133 *

THE CATEGORY NUMBER IS NOT A GUARANTEED INDICATOR OF THE BIAS TO BE EXPECTED ON THE FINAL VALUE OF K-EFFECTIVE
IT SHOULD BE USED WITH CAUTION SINCE MANY OTHER FACTORS ARE INVOLVED (E.G. UNUSUAL/EXOTIC MATERIALS AND NUCLIDES)

1SAMPLING GUIDANCE

1. NO PARTICLE TRACKS IN ZONE 6 FOR 46 STAGES

The first 20 stages that this occurred are:

-1 2 3 4 5 6 10 11 29 30 31 32 33 34 49 52 57 58 60

NO PARTICLE TRACKS IN ZONE 10 FOR 115 STAGES

The first 20 stages that this occurred are:

-1 0 1 2 3 4 5 7 8 10 11 13 14 15 19 20 21 22 23

NO PARTICLE TRACKS IN ZONE 16 FOR 26 STAGES

The first 20 stages that this occurred are:

7 8 15 27 30 32 52 59 60 77 112 114 115 116 117 118 124 128 130

NO PARTICLE TRACKS IN ZONE 21 FOR 68 STAGES

The first 20 stages that this occurred are:

-1 0 2 3 4 5 11 13 15 24 25 26 27 28 29 30 31 32 33

NO PARTICLE TRACKS IN ZONE 26 FOR 86 STAGES

The first 20 stages that this occurred are:

-2 0 1 2 5 7 13 15 16 21 22 24 25 27 28 30 31 33 34

NO PARTICLE TRACKS IN ZONE 78 FOR 146 STAGES

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

NO PARTICLE TRACKS IN ZONE 81 FOR 146 STAGES

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

NO PARTICLE TRACKS IN ZONE 84 FOR 146 STAGES

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

NO PARTICLE TRACKS IN ZONE 87 FOR 146 STAGES

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

NO PARTICLE TRACKS IN ZONE 90 FOR 146 STAGES

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

NO PARTICLE TRACKS IN ZONE 93 FOR 146 STAGES

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 96 FOR 146 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 102 FOR STAGES:
56 59 77 118 122 128 129 130 131 132 134 135 137 140

NO PARTICLE TRACKS IN ZONE 103 FOR 138 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	15	16
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 104 FOR 119 STAGES

The first 20 stages that this occurred are:

-3	-1	0	2	4	5	6	7	8	9	11	12	13	15	16	18	19	20	21
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NO PARTICLE TRACKS IN ZONE 110 FOR 119 STAGES

The first 20 stages that this occurred are:

-3	-2	0	1	2	3	4	5	6	7	8	10	12	13	14	15	17	19	20
----	----	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 115 FOR 139 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 116 FOR 95 STAGES

The first 20 stages that this occurred are:

-1	0	2	4	5	7	8	9	11	16	18	19	20	21	22	24	25	26	29
----	---	---	---	---	---	---	---	----	----	----	----	----	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 117 FOR 138 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 118 FOR 28 STAGES

The first 20 stages that this occurred are:

2	4	5	7	32	52	55	56	58	61	74	76	77	81	117	118	120	122	128
---	---	---	---	----	----	----	----	----	----	----	----	----	----	-----	-----	-----	-----	-----

NO PARTICLE TRACKS IN ZONE 119 FOR 146 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 120 FOR 146 STAGES

The first 20 stages that this occurred are:

-3	-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
----	----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 121 FOR 145 STAGES

The first 20 stages that this occurred are:

-2	-1	0	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	16
----	----	---	---	---	---	---	---	---	---	---	---	----	----	----	----	----	----	----

NO PARTICLE TRACKS IN ZONE 122 FOR 141 STAGES

Figure 6.7-9 MONK8a Output Summary for Connecticut Yankee Fuel Maximum Reactivity
Condition (continued)

The first 20 stages that this occurred are:

-3 -2 -1 0 1 2 3 4 5 6 7 8 9 10 11 12 13 14 15

2. SOURCE PARTICLES STARTED FROM ALL FISSILE ZONES IN ALL STAGES

3. CONSISTENCY OF ESTIMATORS

Individual stage values of K(COLL) and K(SCORE) agree for all stages
Difference between individual stage value of A(SCORE) and 1.0 is greater than 3 SD
Stage : 64
A(SCORE) : 1.02
Difference in SD units: 3.05

4. TEST FOR NORMALITY OF INDIVIDUAL STAGE ESTIMATORS
(Level of significance > 1% for pass)

K(COLL) PASSED Level of significance = 34.79%
K(SCORE) PASSED Level of significance = 8.65%
K(THREE) PASSED Level of significance = 5.57%

5. CHI-SQUARED TEST FOR ADEQUATE SETTLING

	No. of Settling stages	Chi squared per degree of freedom	Probability	
K(COLL)	4	1.088	0.3551	PASSED
K(SCORE)	4	1.325	0.1546	PASSED
K(THREE)	4	1.653	0.0364	PASSED

FINAL VALUE OF K(THREE) = 0.9313 (STDV = 0.0008)

K(THREE) + (3 x STDV) = 0.9337

* MONK PROCESSING COMPLETED TO STAGE 3 *

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel

```
PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
NAC-STC Directly Loaded; Wet Fuel-Pellet Gap; 100% Fuel Geometry Offset; 0 Enric
' Interior Water Density 1 g/cc, Exterior Water Density 0.00001 g/cc
27GROUPNDF4 LATTICECELL
' FUEL
UO2 1 0.95 293 92235 4.5 92238 95.5 END
' CLAD
ZIRCALLOY 2 1 293 END
' H2O CASK INTERIOR
H2O 3 1 293 END
' AL DISK
AL 4 1 293 END
' CASK / DISK STEEL
SS304 5 1 293 END
' BORAL SHEETS
AL 6 DEN=2.6226 0.5738 293 END
B-10 6 DEN=2.6226 0.045 293 END
B-11 6 DEN=2.6226 0.2735 293 END
C 6 DEN=2.6226 0.0926 293 END
' LEAD SHIELD
PB 7 1 293.0 END
' NS4FR SHIELD
B-10 8 0 8.553-5 293 END
B-11 8 0 3.422-4 293 END
AL 8 0 7.763-3 293 END
H 8 0 5.854-2 293 END
O 8 0 2.609-2 293 END
C 8 0 2.264-2 293 END
N 8 0 1.394-3 293 END
' CASK EXTERIOR WATER
H2O 9 0.00001 293 END
' PELLETS CLAD GAP WATER
H2O 10 1 293 END
END COMP
SQUAREPITCH 1.2728 0.8204 1 3 0.9434 2 0.8398 10 END
NAC-STC Directly Loaded; Wet Fuel-Pellet Gap; 100% Fuel Geometry Offset; 0 Enric
READ PARAM RUN=YES PLT=NO TME=5000 GEN=803 NPG=1000 END PARAM
READ GEOM
UNIT 1
COM='FUEL PIN CELL - FOR WATER ELEVATION'
CYLINDER 1 1 0.4102 2P2.3749
CYLINDER 10 1 0.4199 2P2.3749
CYLINDER 2 1 0.4717 2P2.3749
CUBOID 3 1 4P0.6364 2P2.3749
UNIT 2
COM='GUIDE/INSTRUMENT TUBE CELL - FOR WATER ELEVATION'
CYLINDER 3 1 0.5644 2P2.3749
CYLINDER 2 1 0.5992 2P2.3749
CUBOID 3 1 4P0.6364 2P2.3749
UNIT 3
COM='FUEL PIN CELL - FOR STEEL DISK ELEVATION'
CYLINDER 1 1 0.4102 2P0.6350
CYLINDER 10 1 0.4199 2P0.6350
CYLINDER 2 1 0.4717 2P0.6350
CUBOID 3 1 4P0.6364 2P0.6350
UNIT 4
COM='GUIDE/INSTRUMENT TUBE CELL - FOR STEEL DISK ELEVATION'
CYLINDER 3 1 0.5644 2P0.6350
CYLINDER 2 1 0.5992 2P0.6350
CUBOID 3 1 4P0.6364 2P0.6350
UNIT 5
COM='FUEL PIN CELL - FOR AL DISK ELEVATION'
CYLINDER 1 1 0.4102 2P0.7938
CYLINDER 10 1 0.4199 2P0.7938
CYLINDER 2 1 0.4717 2P0.7938
CUBOID 3 1 4P0.6364 2P0.7938
UNIT 6
COM='GUIDE/INSTRUMENT TUBE CELL - FOR AL DISK ELEVATION'
CYLINDER 3 1 0.5644 2P0.7938
CYLINDER 2 1 0.5992 2P0.7938
CUBOID 3 1 4P0.6364 2P0.7938
UNIT 21
COM='ASSEMBLY - FOR WATER ELEVATION'
ARRAY 1 -10.8188 -10.8188 -2.3749
UNIT 22
COM='ASSEMBLY - FOR STEEL DISK ELEVATION'
ARRAY 2 -10.8188 -10.8188 -0.635
UNIT 23
COM='ASSEMBLY - FOR AL DISK ELEVATION'
ARRAY 3 -10.8188 -10.8188 -0.7938
UNIT 31
COM='X-X BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P2.3749
CUBOID 4 1 2P10.3886 2P0.0951 2P2.3749
UNIT 32
COM='Y-Y BORAL SHEET - FOR WATER ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P2.3749
CUBOID 4 1 2P0.0951 2P10.3886 2P2.3749
UNIT 33
COM='X-X BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.6350
CUBOID 4 1 2P10.3886 2P0.0951 2P0.6350
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
UNIT 34
COM='Y-Y BORAL SHEET - FOR STEEL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.6350
CUBOID 4 1 2P0.0951 2P10.3886 2P0.6350
UNIT 35
COM='X-X BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P10.3886 2P0.0635 2P0.7938
CUBOID 4 1 2P10.3886 2P0.0951 2P0.7938
UNIT 36
COM='Y-Y BORAL SHEET - FOR AL DISK ELEVATION'
CUBOID 6 1 2P0.0635 2P10.3886 2P0.7938
CUBOID 4 1 2P0.0951 2P10.3886 2P0.7938
UNIT 40
COM='FUEL TUBE - FOR WATER ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0 -0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 41
COM='FUEL TUBE - FOR WATER ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0 0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 42
COM='FUEL TUBE - FOR WATER ELEVATION (BL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 43
COM='FUEL TUBE - FOR WATER ELEVATION (BR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 44
COM='FUEL TUBE - FOR WATER ELEVATION (TL)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 -0.3698 0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 45
COM='FUEL TUBE - FOR WATER ELEVATION (TR)'
CUBOID 3 1 4P11.1887 2P2.3749
HOLE 21 0.3698 0.3698 0
CUBOID 5 1 4P11.3101 2P2.3749
CUBOID 3 1 4P11.5006 2P2.3749
HOLE 31 0 11.4054 0
HOLE 31 0 -11.4054 0
HOLE 32 11.4054 0 0
HOLE 32 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P2.3749
UNIT 50
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (B)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 51
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (T)'
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0 0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 52
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BL) '
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 53
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (BR) '
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 54
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TL) '
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 -0.3698 0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 55
COM='FUEL TUBE - FOR STEEL DISK ELEVATION (TR) '
CUBOID 3 1 4P11.1887 2P0.6350
HOLE 22 0.3698 0.3698 0
CUBOID 5 1 4P11.3101 2P0.6350
CUBOID 3 1 4P11.5006 2P0.6350
HOLE 33 0 11.4054 0
HOLE 33 0 -11.4054 0
HOLE 34 11.4054 0 0
HOLE 34 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.6350
UNIT 60
COM='FUEL TUBE - FOR AL DISK ELEVATION (B) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0 11.4054 0
HOLE 35 0 -11.4054 0
HOLE 36 11.4054 0 0
HOLE 36 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 61
COM='FUEL TUBE - FOR AL DISK ELEVATION (T) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0 0.3698 0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0 11.4054 0
HOLE 35 0 -11.4054 0
HOLE 36 11.4054 0 0
HOLE 36 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 62
COM='FUEL TUBE - FOR AL DISK ELEVATION (BL) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 -0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0 11.4054 0
HOLE 35 0 -11.4054 0
HOLE 36 11.4054 0 0
HOLE 36 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 63
COM='FUEL TUBE - FOR AL DISK ELEVATION (BR) '
CUBOID 3 1 4P11.1887 2P0.7938
HOLE 23 0.3698 -0.3698 0
CUBOID 5 1 4P11.3101 2P0.7938
CUBOID 3 1 4P11.5006 2P0.7938
HOLE 35 0 11.4054 0
HOLE 35 0 -11.4054 0
HOLE 36 11.4054 0 0
HOLE 36 -11.4054 0 0
CUBOID 5 1 4P11.5461 2P0.7938
UNIT 64
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
COM='FUEL TUBE - FOR AL DISK ELEVATION (TL)'  
CUBOID 3 1 4P11.1887 2P0.7938  
HOLE 23 -0.3698 0.3698 0  
CUBOID 5 1 4P11.3101 2P0.7938  
CUBOID 3 1 4P11.5006 2P0.7938  
HOLE 35 0 11.4054 0  
HOLE 35 0 -11.4054 0  
HOLE 36 11.4054 0 0  
HOLE 36 -11.4054 0 0  
CUBOID 5 1 4P11.5461 2P0.7938  
UNIT 65  
COM='FUEL TUBE - FOR AL DISK ELEVATION (TR)'  
CUBOID 3 1 4P11.1887 2P0.7938  
HOLE 23 0.3698 0.3698 0  
CUBOID 5 1 4P11.3101 2P0.7938  
CUBOID 3 1 4P11.5006 2P0.7938  
HOLE 35 0 11.4054 0  
HOLE 35 0 -11.4054 0  
HOLE 36 11.4054 0 0  
HOLE 36 -11.4054 0 0  
CUBOID 5 1 4P11.5461 2P0.7938  
UNIT 70  
COM='DISK OPENING - FOR WATER ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 40 0 -0.181 0  
UNIT 71  
COM='DISK OPENING - FOR WATER ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 41 0 0.181 0  
UNIT 72  
COM='DISK OPENING - FOR WATER ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 42 -0.181 -0.181 0  
UNIT 73  
COM='DISK OPENING - FOR WATER ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 43 0.181 -0.181 0  
UNIT 74  
COM='DISK OPENING - FOR WATER ELEVATION (TL)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 44 -0.181 0.181 0  
UNIT 75  
COM='DISK OPENING - FOR WATER ELEVATION (TR)'  
CUBOID 3 1 4P11.7272 2P2.3749  
HOLE 45 0.181 0.181 0  
UNIT 80  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 50 0 -0.181 0  
UNIT 81  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 51 0 0.181 0  
UNIT 82  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 52 -0.181 -0.181 0  
UNIT 83  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 53 0.181 -0.181 0  
UNIT 84  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TL)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 54 -0.181 0.181 0  
UNIT 85  
COM='DISK OPENING - FOR STEEL DISK ELEVATION (TR)'  
CUBOID 3 1 4P11.7272 2P0.6350  
HOLE 55 0.181 0.181 0  
UNIT 90  
COM='DISK OPENING - FOR AL DISK ELEVATION (B)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 60 0 -0.181 0  
UNIT 91  
COM='DISK OPENING - FOR AL DISK ELEVATION (T)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 61 0 0.181 0  
UNIT 92  
COM='DISK OPENING - FOR AL DISK ELEVATION (BL)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 62 -0.181 -0.181 0  
UNIT 93  
COM='DISK OPENING - FOR AL DISK ELEVATION (BR)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 63 0.181 -0.181 0  
UNIT 94  
COM='DISK OPENING - FOR AL DISK ELEVATION (TL)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 64 -0.181 0.181 0  
UNIT 95  
COM='DISK OPENING - FOR AL DISK ELEVATION (TR)'  
CUBOID 3 1 4P11.7272 2P0.7938  
HOLE 65 0.181 0.181 0  
UNIT 101  
COM='BASKET STRUCTURE IN TRANSPORT CASK - WATER ELEVATION'
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
CYLINDER 3 1 90.17 2P2.3749
HOLE 75 -58.9864 -40.7822 0
HOLE 75 -58.9864 -13.5941 0
HOLE 73 -58.9864 13.5941 0
HOLE 73 -58.9864 40.7822 0
HOLE 75 -27.1882 -67.9704 0
HOLE 75 -27.1882 -40.7822 0
HOLE 75 -27.1882 -13.5941 0
HOLE 73 -27.1882 13.5941 0
HOLE 73 -27.1882 40.7822 0
HOLE 73 -27.1882 67.9704 0
HOLE 71 0 -67.9704 0
HOLE 71 0 -40.7822 0
HOLE 71 0 -13.5941 0
HOLE 70 0 13.5941 0
HOLE 70 0 40.7822 0
HOLE 70 0 67.9704 0
HOLE 74 27.1882 -67.9704 0
HOLE 74 27.1882 -40.7822 0
HOLE 74 27.1882 -13.5941 0
HOLE 72 27.1882 13.5941 0
HOLE 72 27.1882 40.7822 0
HOLE 72 27.1882 67.9704 0
HOLE 74 58.9864 -40.7822 0
HOLE 74 58.9864 -13.5941 0
HOLE 72 58.9864 13.5941 0
HOLE 72 58.9864 40.7822 0
CYLINDER 5 1 93.98 2P2.3749
CYLINDER 7 1 103.43 2P2.3749
CYLINDER 5 1 110.11 2P2.3749
CYLINDER 9 1 124.12 2P2.3749
CYLINDER 9 1 124.44 2P2.3749
CYLINDER 5 1 125.07 2P2.3749
CUBOID 9 1 4P125.07 2P2.3749
UNIT 102
COM='BASKET STRUCTURE IN TRANSPORT CASK - ST DISK ELEVATION'
CYLINDER 5 1 89.99 2P0.6350
HOLE 85 -58.9864 -40.7822 0
HOLE 85 -58.9864 -13.5941 0
HOLE 83 -58.9864 13.5941 0
HOLE 83 -58.9864 40.7822 0
HOLE 85 -27.1882 -67.9704 0
HOLE 85 -27.1882 -40.7822 0
HOLE 85 -27.1882 -13.5941 0
HOLE 83 -27.1882 13.5941 0
HOLE 83 -27.1882 40.7822 0
HOLE 83 -27.1882 67.9704 0
HOLE 81 0 -67.9704 0
HOLE 81 0 -40.7822 0
HOLE 81 0 -13.5941 0
HOLE 80 0 13.5941 0
HOLE 80 0 40.7822 0
HOLE 80 0 67.9704 0
HOLE 84 27.1882 -67.9704 0
HOLE 84 27.1882 -40.7822 0
HOLE 84 27.1882 -13.5941 0
HOLE 82 27.1882 13.5941 0
HOLE 82 27.1882 40.7822 0
HOLE 82 27.1882 67.9704 0
HOLE 84 58.9864 -40.7822 0
HOLE 84 58.9864 -13.5941 0
HOLE 82 58.9864 13.5941 0
HOLE 82 58.9864 40.7822 0
CYLINDER 3 1 90.17 2P0.6350
CYLINDER 5 1 93.98 2P0.6350
CYLINDER 7 1 103.43 2P0.6350
CYLINDER 5 1 110.11 2P0.6350
CYLINDER 9 1 124.12 2P0.6350
CYLINDER 9 1 124.44 2P0.6350
CYLINDER 5 1 125.07 2P0.6350
CUBOID 9 1 4P125.07 2P0.6350
UNIT 103
COM='BASKET STRUCTURE IN TRANSPORT CASK - AL DISK ELEVATION'
CYLINDER 4 1 89.73 2P0.7938
HOLE 95 -58.9864 -40.7822 0
HOLE 95 -58.9864 -13.5941 0
HOLE 93 -58.9864 13.5941 0
HOLE 93 -58.9864 40.7822 0
HOLE 95 -27.1882 -67.9704 0
HOLE 95 -27.1882 -40.7822 0
HOLE 95 -27.1882 -13.5941 0
HOLE 93 -27.1882 13.5941 0
HOLE 93 -27.1882 40.7822 0
HOLE 93 -27.1882 67.9704 0
HOLE 91 0 -67.9704 0
HOLE 91 0 -40.7822 0
HOLE 91 0 -13.5941 0
HOLE 90 0 13.5941 0
HOLE 90 0 40.7822 0
HOLE 90 0 67.9704 0
HOLE 94 27.1882 -67.9704 0
HOLE 94 27.1882 -40.7822 0
HOLE 94 27.1882 -13.5941 0
HOLE 92 27.1882 13.5941 0
HOLE 92 27.1882 40.7822 0
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
HOLE 92 27.1882 67.9704 0
HOLE 94 58.9864 -40.7822 0
HOLE 94 58.9864 -13.5941 0
HOLE 92 58.9864 13.5941 0
HOLE 92 58.9864 40.7822 0
CYLINDER 3 1 90.17 2P0.7938
CYLINDER 5 1 93.98 2P0.7938
CYLINDER 7 1 103.43 2P0.7938
CYLINDER 5 1 110.11 2P0.7938
CYLINDER 9 1 124.12 2P0.7938
CYLINDER 9 1 124.44 2P0.7938
CYLINDER 5 1 125.07 2P0.7938
CUBOID 9 1 4P125.07 2P0.7938
GLOBAL UNIT 104
COM='DISK SLICE STACK'
ARRAY 4 -125.07 -125.07 0
CUBOID 9 1 4P125.08 12.3573 0
END GEOM
READ ARRAY
ARA=1 NUX=17 NUY=17 NUZ=1 FILL
34R1
5R1 2 2R1 2 2R1 2 5R1
3R1 2 9R1 2 3R1
17R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
34R1
2R1 2 2R1 2 2R1 2 2R1 2 2R1
17R1
3R1 2 9R1 2 3R1
5R1 2 2R1 2 2R1 2 5R1
34R1
END FILL
ARA=2 NUX=17 NUY=17 NUZ=1 FILL
34R3
5R3 4 2R3 4 2R3 4 5R3
3R3 4 9R3 4 3R3
17R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3
34R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3
34R3
2R3 4 2R3 4 2R3 4 2R3 4 2R3
17R3
3R3 4 9R3 4 3R3
5R3 4 2R3 4 2R3 4 5R3
34R3
END FILL
ARA=3 NUX=17 NUY=17 NUZ=1 FILL
34R5
5R5 6 2R5 6 2R5 6 5R5
3R5 6 9R5 6 3R5
17R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5
34R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5
34R5
2R5 6 2R5 6 2R5 6 2R5 6 2R5
17R5
3R5 6 9R5 6 3R5
5R5 6 2R5 6 2R5 6 5R5
34R5
END FILL
ARA=4 NUX=1 NUY=1 NUZ=4 FILL 101 102 101 103 END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIRROR END BOUNDS
READ PLOT
TTL='XY SLICE OF CASK - ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-120.0 YUL=120.0 ZUL=5.5 XLR=120.0 YLR=-120.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='XY SLICE CASK CENTER AREA ST DISK ELEVATION'
SCR=YES PIC=MAT LPI=10
XUL=-27.0 YUL=27.0 ZUL=5.5 XLR=27.0 YLR=-27.0 ZLR=5.5
UAX=1.0 VDN=-1.0 NAX=1500 END
END PLOT
END DATA
```

SECONDARY MODULE 000008 HAS BEEN CALLED.

MODULE 000008 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 1.37 (SECONDS).

SECONDARY MODULE 000002 HAS BEEN CALLED.

MODULE 000002 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 6.49 (SECONDS).

SECONDARY MODULE 000009 HAS BEEN CALLED.

MODULE 000009 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 672.94 (SECONDS).

MODULE CSAS25 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 686.79 (SECONDS).

THE FOLLOWING DATA CARDS PRECEDE AN = CARD
EXECUTION TERMINATED DUE TO ERRORS

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

**** PROBLEM PARAMETERS ****

LIB 27GROUPNDF4  LIBRARY
MX 10 MIXTURES
MSC 19 COMPOSITION SPECIFICATIONS
IZM 4 MATERIAL ZONES
GE LATTICECELL  GEOMETRY
MORE 0 0/1 DO NOT READ/READ OPTIONAL PARAMETER DATA
MSLN 0 FUEL SOLUTIONS

**** PROBLEM COMPOSITION DESCRIPTION ****

SC UO2          STANDARD COMPOSITION
MX 1 MIXTURE NO.
VF 0.9500 VOLUME FRACTION
ROTH 10.9600 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      92000      1.00 ATOM/MOLECULE
                  92235      4.500 WT%
                  92238      95.500 WT%
      8016      2.00 ATOMS/MOLECULE

'CLAD
END

SC ZIRCALLOY    STANDARD COMPOSITION
MX 2 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 6.5600 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      40302      1.00 ATOM/MOLECULE

'H2O CASK INTERIOR
END

SC H2O          STANDARD COMPOSITION
MX 3 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 0.9982 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      1001      2.00 ATOMS/MOLECULE
      8016      1.00 ATOM/MOLECULE

'AL DISK
END

SC AL           STANDARD COMPOSITION
MX 4 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 2.7020 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      13027      1.00 ATOM/MOLECULE

'CASK / DISK STEEL
END

SC SS304        STANDARD COMPOSITION
MX 5 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 7.9200 THEORETICAL DENSITY
NEL 4 NO. ELEMENTS
ICP 0 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      24304      19.000 WT%
      25055      2.000 WT%
      26304      69.500 WT%
      28304      9.500 WT%

'BORAL SHEETS
END

SC AL           STANDARD COMPOSITION
MX 6 MIXTURE NO.
VF 0.5738 VOLUME FRACTION
ROTH 2.6226 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
      13027      1.00 ATOM/MOLECULE
END

SC B-10         STANDARD COMPOSITION
MX 6 MIXTURE NO.
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
VF      0.0450 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5010      1.00 ATOM/MOLECULE
END

SC B-11      STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF      0.2735 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5011      1.00 ATOM/MOLECULE
END

SC C          STANDARD COMPOSITION
MX          6 MIXTURE NO.
VF      0.0926 VOLUME FRACTION
ROTH    2.6226 SPECIFIED DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        6012      1.00 ATOM/MOLECULE

'LEAD SHIELD
END

SC PB        STANDARD COMPOSITION
MX          7 MIXTURE NO.
VF      1.0000 VOLUME FRACTION
ROTH    11.3440 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        82000     1.00 ATOM/MOLECULE

'NS4FR SHIELD
END

SC B-10      STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     8.5530E-05 ATOMIC DENSITY
ROTH    1.0000 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5010      1.00 ATOM/MOLECULE
END

SC B-11      STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     3.4220E-04 ATOMIC DENSITY
ROTH    1.0000 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        5011      1.00 ATOM/MOLECULE
END

SC AL        STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     7.7630E-03 ATOMIC DENSITY
ROTH    2.7020 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        13027     1.00 ATOM/MOLECULE
END

SC H          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     5.8540E-02 ATOMIC DENSITY
ROTH    1.0000 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        1001      1.00 ATOM/MOLECULE
END

SC O          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     2.6090E-02 ATOMIC DENSITY
ROTH    1.0000 THEORETICAL DENSITY
NEL      1 NO. ELEMENTS
ICP      1 0/1 MIXTURE/COMPOUND
TEMP    293.0 DEG KELVIN
        8016      1.00 ATOM/MOLECULE
END

SC C          STANDARD COMPOSITION
MX          8 MIXTURE NO.
DEN     2.2640E-02 ATOMIC DENSITY
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
ROTH      2.1000 THEORETICAL DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN

        6012      1.00 ATOM/MOLECULE
END

SC N      STANDARD COMPOSITION
MX        8 MIXTURE NO.
DEN      1.3940E-03 ATOMIC DENSITY
ROTH      1.0000 THEORETICAL DENSITY
NEL        1 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
        7014      1.00 ATOM/MOLECULE

'CASK EXTERIOR WATER
END

SC H2O     STANDARD COMPOSITION
MX        9 MIXTURE NO.
VF        0.0000 VOLUME FRACTION
ROTH      0.9982 THEORETICAL DENSITY
NEL        2 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
        1001      2.00 ATOMS/MOLECULE
        8016      1.00 ATOM/MOLECULE

'PELLET CLAD GAP WATER
END

SC H2O     STANDARD COMPOSITION
MX        10 MIXTURE NO.
VF        1.0000 VOLUME FRACTION
ROTH      0.9982 THEORETICAL DENSITY
NEL        2 NO. ELEMENTS
ICP        1 0/1 MIXTURE/COMPOUND
TEMP      293.0 DEG KELVIN
        1001      2.00 ATOMS/MOLECULE
        8016      1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH     1.2728 CM CENTER TO CENTER SPACING
FUELOD    0.8204 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL      1 MIXTURE NO. OF FUEL
MMOD       3 MIXTURE NO. OF MODERATOR
CLADOD    0.9434 CM CLAD OUTER DIAMETER
MCLAD      2 MIXTURE NO. OF CLAD
GAPOD     0.8398 CM GAP OUTER DIAMETER
MGAP       10 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

        ZONE 1 IS FUEL
        ZONE 2 IS GAP
        ZONE 3 IS CLAD
        ZONE 4 IS MOD

LOGICAL ASSIGNMENTS

MASTER LIBRARY 11
WORKING LIBRARY 0
SCRATCH FILE   18
NEW LIBRARY     1

P R O B L E M   D E S C R I P T I O N

IGR--GEOMETRY (0/1/2/3--INF MED/SLAB/CYL/SPHERE)      2
IZM--NUMBER OF ZONES OR MATERIAL REGIONS              10
MS--MIXING TABLE LENGTH                              27
IBL--SHIELDED CROSS SECTION EDIT OPTION (0/1--NO/YES) 0
IBR--BONDARENKO FACTOR EDIT OPTION (0/1--NO/YES)      0
ISSOPT--DANCOFF FACTOR OPTION                        00
CONVERGENCE CRITERION      1.00000E-03
GEOMETRY CORRECTION FACTOR FOR WIGNER RATIONAL APPROXIMATION 1.350E+00

        3Q ARRAY HAS      27 ENTRIES.
        4Q ARRAY HAS      27 ENTRIES.
        5Q ARRAY HAS      27 ENTRIES.
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```

6Q ARRAY HAS      10 ENTRIES.
7Q ARRAY HAS      10 ENTRIES.
8Q ARRAY HAS      10 ENTRIES.
9Q ARRAY HAS      10 ENTRIES.
10Q ARRAY HAS     27 ENTRIES.
11Q ARRAY HAS     10 ENTRIES.

MIXING TABLE
ENTRY MIXTURE ISOTOPE NUMBER DENSITY NEW IDENTIFIER
1 1 92235 1.05821E-03 1092235
2 1 92238 2.21739E-02 1092238
3 1 8016 4.64643E-02 1008016
4 3 8016 3.33846E-02 3008016
5 8 8016 2.60900E-02 8008016
6 9 8016 3.33846E-07 9008016
7 10 8016 3.33846E-02 10008016
8 2 40302 4.33078E-02 2040302
9 3 1001 6.67692E-02 3001001
10 8 1001 5.85400E-02 8001001
11 9 1001 6.67692E-07 9001001
12 10 1001 6.67692E-02 10001001
13 4 13027 6.03066E-02 4013027
14 6 13027 3.35871E-02 6013027
15 8 13027 7.76300E-03 8013027
16 5 24304 1.74286E-02 5024304
17 5 25055 1.73633E-03 5025055
18 5 26304 5.93579E-02 5026304
19 5 28304 7.72070E-03 5028304
20 6 5010 7.09799E-03 6005010
21 8 5010 8.55300E-05 8005010
22 6 5011 3.92356E-02 6005011
23 8 5011 3.42200E-04 8005011
24 6 6012 1.21874E-02 6006012
25 8 6012 2.26400E-02 8006012
26 7 82000 3.29690E-02 7082000
27 8 7014 1.39400E-03 8007014

GEOMETRY AND MATERIAL DESCRIPTION
ZONE MIXTURE OUTER DIMENSION TEMPERATURE EXTRA XS TYPE (0/1--FUEL/MOD)
1 1 4.10200E-01 2.93000E+02 1.23440E+00 0
2 10 4.19900E-01 2.93000E+02 0.00000E+00 0
3 2 4.71700E-01 2.93000E+02 6.63851E+00 0
4 3 7.18100E-01 2.93000E+02 0.00000E+00 0
5 4 5.71810E+00 2.93000E+02 0.00000E+00 0
6 5 1.07181E+01 2.93000E+02 0.00000E+00 0
7 6 1.57181E+01 2.93000E+02 0.00000E+00 0
8 7 2.07181E+01 2.93000E+02 0.00000E+00 0
9 8 2.57181E+01 2.93000E+02 0.00000E+00 0
10 9 3.07181E+01 2.93000E+02 0.00000E+00 0

*****
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC**
*****
NUMERIC PARAMETERS
*****
TME MAXIMUM PROBLEM TIME (MIN) *****
TBA TIME PER GENERATION (MIN) 0.50
GEN NUMBER OF GENERATIONS 803
NPG NUMBER PER GENERATION 1000
NSK NUMBER OF GENERATIONS TO BE SKIPPED 3
BEG BEGINNING GENERATION NUMBER 1
RES GENERATIONS BETWEEN CHECKPOINTS 0
X1D NUMBER OF EXTRA 1-D CROSS SECTIONS 1
NBK NEUTRON BANK SIZE 1025
XNB EXTRA POSITIONS IN NEUTRON BANK 0
NFB FISSION BANK SIZE 1000
XFB EXTRA POSITIONS IN FISSION BANK 0
WTA DEFAULT VALUE OF WEIGHT AVERAGE 0.5000
WTH WEIGHT HIGH FOR SPLITTING 3.0000
WTL WEIGHT LOW FOR RUSSIAN ROULETTE 0.3333
RND STARTING RANDOM NUMBER BB827100001
NB8 NUMBER OF D.A. BLOCKS ON UNIT 8 200
NL8 LENGTH OF D.A. BLOCKS ON UNIT 8 512
ADJ MODE OF CALCULATION FORWARD
INPUT DATA WRITTEN ON RESTART UNIT NO
BINARY DATA INTERFACE YES

```

```

*****
***                                     NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC****
***
***** LOGICAL PARAMETERS *****
***
*** RUN EXECUTE PROBLEM AFTER CHECKING DATA YES PLT PLOT PICTURE MAP(S) NO ***
*** FLX COMPUTE FLUX NO PDN COMPUTE FISSION DENSITIES NO ***
*** SMU COMPUTE AVG UNIT SELF-MULTIPLICATION NO NUB COMPUTE NU-BAR & AVG FISSION GROUP YES ***
*** MKU COMPUTE MATRIX K-EFF BY UNIT NUMBER NO MKP COMPUTE MATRIX K-EFF BY UNIT LOCATION NO ***
*** CKU COMPUTE COFACTOR K-EFF BY UNIT NUMBER NO CKP COMPUTE COFACTOR K-EFF BY UNIT LOCATION NO ***
*** FMU PRINT FISS PROD MATRIX BY UNIT NUMBER NO FMP PRINT FISS PROD MATRIX BY UNIT LOCATION NO ***
*** MKH COMPUTE MATRIX K-EFF BY HOLE NUMBER NO MKA COMPUTE MATRIX K-EFF BY ARRAY NUMBER NO ***
*** CKH COMPUTE COFACTOR K-EFF BY HOLE NUMBER NO CKA COMPUTE COFACTOR K-EFF BY ARRAY NUMBER NO ***
*** FMH PRINT FISS PROD MATRIX BY HOLE NUMBER NO FMA PRINT FISS PROD MATRIX BY ARRAY NUMBER NO ***
*** HHL COLLECT MATRIX BY HIGHEST HOLE LEVEL NO HAL COLLECT MATRIX BY HIGHEST ARRAY LEVEL NO ***
*** AMX PRINT ALL MIXED CROSS SECTIONS NO FAR PRINT FIS. AND ABS. BY REGION NO ***
*** XS1 PRINT 1-D MIXTURE X-SECTIONS NO GAS PRINT FAR BY GROUP NO ***
*** XS2 PRINT 2-D MIXTURE X-SECTIONS NO PAX PRINT XSEC-ALBEDO CORRELATION TABLES NO ***
*** XAP PRINT MIXTURE ANGLES & PROBABILITIES NO PWT PRINT WEIGHT AVERAGE ARRAY NO ***
*** PKI PRINT FISSION SPECTRUM NO PGM PRINT INPUT GEOMETRY NO ***
*** P1D PRINT EXTRA 1-D CROSS SECTIONS NO BUG PRINT DEBUG INFORMATION NO ***
*** TRK PRINT TRACKING INFORMATION NO ***
***
*****
*****
PARAMETER INPUT COMPLETED

```

```
..... 0 IO'S WERE USED READING THE PARAMETER DATA .....
```

***** DATA READING COMPLETED *****
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

MIXING TABLE

NUMBER OF SCATTERING ANGLES = 2
CROSS SECTION MESSAGE THRESHOLD = 3.0E-05

6.7-279

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

5026304	5.93579E-02	6.95000E-01	26000	55.8447	FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED
08/12/94						
5028304	7.72070E-03	9.50001E-02	28000	58.6872	NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED
08/12/94						
MIXTURE = 6 DENSITY(G/CC) = 2.5830						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
6005010	7.09799E-03	4.56901E-02	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
6005011	3.92356E-02	2.77698E-01	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
6006012	1.21874E-02	9.40196E-02	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
6013027	3.35871E-02	5.82592E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE = 7 DENSITY(G/CC) = 11.344						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
7082000	3.29690E-02	1.00000E+00	82000	207.2100	PB 1288 218NGP 042375 P-3 293K	UPDATED
08/12/94						
MIXTURE = 8 DENSITY(G/CC) = 1.6298						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
8001001	5.85400E-02	6.01023E-02	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
8005010	8.55300E-05	8.72589E-04	5010	10.0130	B-10 1273 218NGP 042375 P-3 293K	UPDATED
08/12/94						
8005011	3.42200E-04	3.83863E-03	5011	11.0096	BORON-11 ENDF/B-IV MAT 1160	UPDATED
08/12/94						
8006012	2.26400E-02	2.76813E-01	6000	12.0001	CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED
08/12/94						
8007014	1.39400E-03	1.98893E-02	7014	14.0033	NITROGEN-14 ENDF/B-IV MAT 1275	UPDATED
08/12/94						
8008016	2.60900E-02	4.25068E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
8013027	7.76300E-03	2.13416E-01	13027	26.9818	AL-27 1193 218 GP 040375(5)	UPDATED
08/12/94						
MIXTURE = 9 DENSITY(G/CC) = 0.99817E-05						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
9001001	6.67692E-07	1.11927E-01	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
9008016	3.33846E-07	8.88073E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
MIXTURE = 10 DENSITY(G/CC) = 0.99817						
NUCLIDE	ATOM-DENS.	WGT. FRAC.	ZA	AWT	NUCLIDE TITLE	
10001001	6.67692E-01	1.11927E-01	1001	1.0077	HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED
08/12/94						
10008016	3.33846E-02	8.88074E-01	8016	15.9904	OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED
08/12/94						
3001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
8001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
9001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
10001001					HYDROGEN ENDF/B-IV MAT 1269/THRM1002	UPDATED 08/12/94
6005010					B-10 1273 218NGP 042375 P-3 293K	UPDATED 08/12/94
8005010					B-10 1273 218NGP 042375 P-3 293K	UPDATED 08/12/94
6005011					BORON-11 ENDF/B-IV MAT 1160	UPDATED 08/12/94
8005011					BORON-11 ENDF/B-IV MAT 1160	UPDATED 08/12/94
6006012					CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
8006012					CARBON-12 ENDF/B-IV MAT 1274/THRM1065	UPDATED 08/12/94
8007014					NITROGEN-14 ENDF/B-IV MAT 1275	UPDATED 08/12/94
1008016					OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED 08/12/94
3008016					OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED 08/12/94
8008016					OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED 08/12/94
9008016					OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED 08/12/94
10008016					OXYGEN-16 ENDF/B-IV MAT 1276	UPDATED 08/12/94
4013027					AL-27 1193 218 GP 040375(5)	UPDATED 08/12/94
6013027					AL-27 1193 218 GP 040375(5)	UPDATED 08/12/94
8013027					AL-27 1193 218 GP 040375(5)	UPDATED 08/12/94
5024304					CR 1191 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94
5025055					MANGANESE-55 ENDF/B-IV MAT 1197	UPDATED 08/12/94
5026304					FE 1192 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94
5028304					NI 1190 WT SS-304(1/EST) P-3 293K SP=5+4(42375)	UPDATED 08/12/94
2040302					ZIRCALLOY ENDF/B-IV MAT 1284	UPDATED 08/12/94
7082000					PB 1288 218NGP 042375 P-3 293K	UPDATED 08/12/94
1092235					URANIUM-235 ENDF/B-IV MAT 1261	UPDATED 08/12/94
1092238					URANIUM-238 ENDF/B-IV MAT 1262	UPDATED 08/12/94

KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 3 WERE CORRECTED FOR BAD MOMENTS.
KENO MESSAGE NUMBER K5-222 2 TRANSFERS FOR MIXTURE 9 WERE CORRECTED FOR BAD MOMENTS.
KENO MESSAGE NUMBER K5-222 1 TRANSFERS FOR MIXTURE 10 WERE CORRECTED FOR BAD MOMENTS.

..... 0 IO'S WERE USED MIXING CROSS-SECTIONS

1-D CROSS SECTION ARRAY ID NUMBERS
1 2002 1452 27 18 1018

..... 0 IO'S WERE USED PREPARING THE CROSS SECTIONS

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```

*****
***      NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC      ***
***                                                                                               ***
*****
***                               ***** ADDITIONAL INFORMATION *****                               ***
***                                                                                               ***
*** NUMBER OF ENERGY GROUPS          27      USE LATTICE GEOMETRY                               YES ***
*** NO. OF FISSION SPECTRUM SOURCE GROUP 1      GLOBAL ARRAY NUMBER                               4 ***
*** NO. OF SCATTERING ANGLES IN XSECS    2      NUMBER OF UNITS IN THE GLOBAL X DIR.          1 ***
*** ENTRIES/NEUTRON IN THE NEUTRON BANK 25      NUMBER OF UNITS IN THE GLOBAL Y DIR.          1 ***
*** ENTRIES/NEUTRON IN THE FISSION BANK 18      NUMBER OF UNITS IN THE GLOBAL Z DIR.          4 ***
*** NUMBER OF MIXTURES USED              9      USE A GLOBAL REFLECTOR                               YES ***
*** NUMBER OF BIAS ID'S USED              1      USE NESTED HOLES                               YES ***
*** NUMBER OF DIFFERENTIAL ALBEDOS USED    0      NUMBER OF HOLES                               186 ***
*** TOTAL INPUT GEOMETRY REGIONS          154      MAXIMUM HOLE NESTING LEVEL                3 ***
*** NUMBER OF GEOMETRY REGIONS USED        154      USE NESTED ARRAYS                               YES ***
*** LARGEST GEOMETRY UNIT NUMBER           104      NUMBER OF ARRAYS USED                               4 ***
*** LARGEST ARRAY NUMBER                    4      MAXIMUM ARRAY NESTING LEVEL                2 ***
***                                                                                               ***
*** +X BOUNDARY CONDITION          MIRROR      -X BOUNDARY CONDITION          MIRROR ***
*** +Y BOUNDARY CONDITION          MIRROR      -Y BOUNDARY CONDITION          MIRROR ***
*** +Z BOUNDARY CONDITION          PER         -Z BOUNDARY CONDITION          PER ***
***                                                                                               ***
*****

```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```

*****
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENR
*****

***** SPACE AND SUPERGROUP INFORMATION *****
100000 WORDS IS THE TOTAL SPACE AVAILABLE.
53021 WORDS WERE USED FOR NON-SUPERGROUP STORAGE.
46979 WORDS OF STORAGE ARE AVAILABLE FOR SUPERGROUPED DATA.
99620 WORDS OF STORAGE ARE AVAILABLE FOR CONSTRUCTING THE SUPERGROUPS.
46919 WORDS OF STORAGE ARE AVAILABLE TO EACH SUPERGROUP.
1180 WORDS ARE NEEDED FOR THE LARGEST GROUP.
54417 WORDS OF STORAGE IS SUFFICIENT TO RUN THIS PROBLEM.
65927 WORDS OF STORAGE WILL ALLOW THE PROBLEM TO RUN WITH ONE SUPERGROUP.
66208 WORDS OF STORAGE WILL BE USED TO RUN THIS PROBLEM.
*****

*****
SUPERGROUP      STARTING      ENDING      XSEC      ALBEDO      TOTAL
                  GROUP        GROUP      LENGTH    LENGTH      LENGTH
*****
          1          1          27        2523         0        12846
*****

..... 0 IO'S WERE USED IN SUPERGROUPING .....

*****
**  ARRAY      UNITS IN  UNITS IN  UNITS IN  NESTING  **
**  NUMBER      X DIR.   Y DIR.   Z DIR.   LEVEL   **
**  1           17      17          1        2      **
**  2           17      17          1        2      **
**  3           17      17          1        2      **
**  4 GLOBAL     1       1           4        1      **
*****

..... 0 IO'S WERE USED LOADING THE DATA .....

```


Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

GENERATION	K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	922 INDEPENDENT	FISSION POINTS WERE GENERATED			
1	8.42775E-01	4.89833E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING...ONLY	973 INDEPENDENT	FISSION POINTS WERE GENERATED			
2	8.59256E-01	5.03500E-01	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
3	9.81674E-01	5.17333E-01	9.81674E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	9.36266E-01	5.32000E-01	9.58970E-01	2.27041E-02	0.00000E+00	0.00000E+00
5	9.32953E-01	5.46500E-01	9.50297E-01	1.57173E-02	0.00000E+00	0.00000E+00
6	9.22000E-01	5.60333E-01	9.43223E-01	1.31743E-02	0.00000E+00	0.00000E+00
7	8.90982E-01	5.75000E-01	9.32775E-01	1.46049E-02	0.00000E+00	0.00000E+00
8	9.58241E-01	5.88667E-01	9.37019E-01	1.26576E-02	0.00000E+00	0.00000E+00
9	9.58604E-01	6.02333E-01	9.40103E-01	1.11332E-02	0.00000E+00	0.00000E+00
10	9.31171E-01	6.16167E-01	9.38986E-01	9.70606E-03	0.00000E+00	0.00000E+00
11	9.14921E-01	6.29833E-01	9.36312E-01	8.96786E-03	0.00000E+00	0.00000E+00
12	8.89081E-01	6.43667E-01	9.31589E-01	9.30837E-03	0.00000E+00	0.00000E+00
13	8.96803E-01	6.58333E-01	9.28427E-01	8.99403E-03	0.00000E+00	0.00000E+00
14	9.29014E-01	6.72000E-01	9.28476E-01	8.21053E-03	0.00000E+00	0.00000E+00
15	9.17850E-01	6.85667E-01	9.27658E-01	7.59669E-03	0.00000E+00	0.00000E+00
16	9.67038E-01	6.99500E-01	9.30471E-01	7.57478E-03	0.00000E+00	0.00000E+00
17	9.04308E-01	7.13167E-01	9.28727E-01	7.26425E-03	0.00000E+00	0.00000E+00
18	8.82455E-01	7.26833E-01	9.25835E-01	7.38491E-03	0.00000E+00	0.00000E+00
19	9.23019E-01	7.40667E-01	9.25669E-01	6.93889E-03	0.00000E+00	0.00000E+00
20	9.26944E-01	7.54333E-01	9.25740E-01	6.54243E-03	0.00000E+00	0.00000E+00
21	9.26825E-01	7.68167E-01	9.25797E-01	6.18878E-03	0.00000E+00	0.00000E+00
22	9.51700E-01	7.81833E-01	9.27092E-01	6.01235E-03	0.00000E+00	0.00000E+00
23	9.31857E-01	7.95500E-01	9.27319E-01	5.72338E-03	0.00000E+00	0.00000E+00
24	9.45504E-01	8.08333E-01	9.28146E-01	5.51927E-03	0.00000E+00	0.00000E+00
25	9.48368E-01	8.23000E-01	9.29025E-01	5.34664E-03	0.00000E+00	0.00000E+00
26	9.29654E-01	8.36833E-01	9.29051E-01	5.11908E-03	0.00000E+00	0.00000E+00
27	8.95124E-01	8.50500E-01	9.27694E-01	5.09414E-03	0.00000E+00	0.00000E+00
28	9.51964E-01	8.63333E-01	9.28628E-01	4.98251E-03	0.00000E+00	0.00000E+00
29	9.46722E-01	8.77000E-01	9.29298E-01	4.84104E-03	0.00000E+00	0.00000E+00
30	9.07429E-01	8.89833E-01	9.28517E-01	4.72987E-03	0.00000E+00	0.00000E+00
31	9.19008E-01	9.03667E-01	9.28189E-01	4.57562E-03	0.00000E+00	0.00000E+00
32	9.23612E-01	9.16333E-01	9.28036E-01	4.42310E-03	0.00000E+00	0.00000E+00
33	9.32488E-01	9.31000E-01	9.28180E-01	4.28045E-03	0.00000E+00	0.00000E+00
34	9.03881E-01	9.43833E-01	9.27421E-01	4.21352E-03	0.00000E+00	0.00000E+00
35	9.62378E-01	9.56667E-01	9.28480E-01	4.21899E-03	0.00000E+00	0.00000E+00
36	9.97305E-01	9.70333E-01	9.27563E-01	4.19447E-03	0.00000E+00	0.00000E+00
37	9.73499E-01	9.83167E-01	9.28875E-01	4.27911E-03	0.00000E+00	0.00000E+00
38	9.15334E-01	9.97000E-01	9.28499E-01	4.17552E-03	0.00000E+00	0.00000E+00
39	9.46079E-01	1.01067E+00	9.28974E-01	4.08880E-03	0.00000E+00	0.00000E+00
40	9.66997E-01	1.02450E+00	9.29975E-01	4.10360E-03	0.00000E+00	0.00000E+00
41	9.39366E-01	1.03717E+00	9.30216E-01	4.00424E-03	0.00000E+00	0.00000E+00
42	9.67038E-01	1.05000E+00	9.31136E-01	4.00995E-03	0.00000E+00	0.00000E+00
768	9.46328E-01	1.07178E+01	9.33891E-01	8.50544E-04	0.00000E+00	0.00000E+00
769	9.59888E-01	1.07317E+01	9.33925E-01	8.50111E-04	0.00000E+00	0.00000E+00
770	9.59446E-01	1.07462E+01	9.33958E-01	8.49653E-04	0.00000E+00	0.00000E+00
771	9.79919E-01	1.07600E+01	9.34018E-01	8.50650E-04	0.00000E+00	0.00000E+00
772	9.06141E-01	1.07728E+01	9.33981E-01	8.50315E-04	0.00000E+00	0.00000E+00
773	9.42108E-01	1.07865E+01	9.33992E-01	8.49277E-04	0.00000E+00	0.00000E+00
774	9.46554E-01	1.07993E+01	9.34008E-01	8.48332E-04	0.00000E+00	0.00000E+00
775	9.31532E-01	1.08122E+01	9.34005E-01	8.47240E-04	0.00000E+00	0.00000E+00
776	9.55431E-01	1.08258E+01	9.34033E-01	8.46598E-04	0.00000E+00	0.00000E+00
777	9.00955E-01	1.08387E+01	9.33990E-01	8.46581E-04	0.00000E+00	0.00000E+00
778	9.18847E-01	1.08525E+01	9.33970E-01	8.45715E-04	0.00000E+00	0.00000E+00
779	9.18019E-01	1.08652E+01	9.33950E-01	8.44875E-04	0.00000E+00	0.00000E+00
780	9.58540E-01	1.08780E+01	9.33981E-01	8.44380E-04	0.00000E+00	0.00000E+00
781	9.33157E-01	1.08900E+01	9.33980E-01	8.43296E-04	0.00000E+00	0.00000E+00
782	9.41474E-01	1.09037E+01	9.33990E-01	8.42269E-04	0.00000E+00	0.00000E+00
783	9.81031E-01	1.09157E+01	9.34050E-01	8.43344E-04	0.00000E+00	0.00000E+00
784	9.34474E-01	1.09302E+01	9.34051E-01	8.42265E-04	0.00000E+00	0.00000E+00
785	9.61076E-01	1.09430E+01	9.34085E-01	8.41896E-04	0.00000E+00	0.00000E+00
786	9.36456E-01	1.09568E+01	9.34088E-01	8.40827E-04	0.00000E+00	0.00000E+00
787	9.33709E-01	1.09705E+01	9.34088E-01	8.39755E-04	0.00000E+00	0.00000E+00
788	9.36210E-01	1.09842E+01	9.34091E-01	8.38690E-04	0.00000E+00	0.00000E+00
789	9.18046E-01	1.09980E+01	9.34070E-01	8.37872E-04	0.00000E+00	0.00000E+00
790	9.03295E-01	1.10117E+01	9.34031E-01	8.37719E-04	0.00000E+00	0.00000E+00
791	9.24554E-01	1.10255E+01	9.34019E-01	8.36743E-04	0.00000E+00	0.00000E+00
792	9.63797E-01	1.10392E+01	9.34057E-01	8.36533E-04	0.00000E+00	0.00000E+00
793	9.54726E-01	1.10520E+01	9.34083E-01	8.35883E-04	0.00000E+00	0.00000E+00
794	9.23635E-01	1.10657E+01	9.34070E-01	8.34931E-04	0.00000E+00	0.00000E+00
795	9.41438E-01	1.10785E+01	9.34079E-01	8.33929E-04	0.00000E+00	0.00000E+00
796	8.87331E-01	1.10932E+01	9.34020E-01	8.34957E-04	0.00000E+00	0.00000E+00
797	9.41992E-01	1.11060E+01	9.34030E-01	8.33966E-04	0.00000E+00	0.00000E+00
798	9.52630E-01	1.11197E+01	9.34054E-01	8.33246E-04	0.00000E+00	0.00000E+00
799	9.12950E-01	1.11343E+01	9.34027E-01	8.32621E-04	0.00000E+00	0.00000E+00
800	9.10958E-01	1.11482E+01	9.33998E-01	8.32079E-04	0.00000E+00	0.00000E+00
801	9.14560E-01	1.11610E+01	9.33974E-01	8.31393E-04	0.00000E+00	0.00000E+00
802	9.42908E-01	1.11747E+01	9.33985E-01	8.30428E-04	0.00000E+00	0.00000E+00
803	8.95273E-01	1.11883E+01	9.33937E-01	8.30798E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123

EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

LIFETIME = 3.80463E-05 + OR - 7.45568E-08 GENERATION TIME = 2.68138E-05 + OR - 3.95380E-08
 NU BAR = 2.43935E+00 + OR - 6.92015E-05 AVERAGE FISSION GROUP = 2.19694E+01 + OR - 4.03476E-03
 ENERGY(EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 2.33262E-01 + OR - 7.60910E-04

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.93388	+ OR - 0.00083	0.93305 TO 0.93471	0.93222 TO 0.93554	0.93139 TO 0.93637	800000
4	0.93387	+ OR - 0.00083	0.93304 TO 0.93470	0.93221 TO 0.93554	0.93138 TO 0.93637	799000
5	0.93388	+ OR - 0.00083	0.93304 TO 0.93471	0.93221 TO 0.93554	0.93138 TO 0.93637	798000
6	0.93389	+ OR - 0.00083	0.93306 TO 0.93472	0.93222 TO 0.93556	0.93139 TO 0.93639	797000
7	0.93394	+ OR - 0.00083	0.93311 TO 0.93478	0.93228 TO 0.93561	0.93145 TO 0.93644	796000
8	0.93391	+ OR - 0.00083	0.93308 TO 0.93475	0.93225 TO 0.93558	0.93142 TO 0.93641	795000
9	0.93388	+ OR - 0.00083	0.93305 TO 0.93472	0.93222 TO 0.93555	0.93138 TO 0.93638	794000
10	0.93389	+ OR - 0.00083	0.93305 TO 0.93472	0.93222 TO 0.93555	0.93138 TO 0.93639	793000
11	0.93391	+ OR - 0.00083	0.93308 TO 0.93474	0.93224 TO 0.93558	0.93141 TO 0.93641	792000
12	0.93397	+ OR - 0.00083	0.93313 TO 0.93480	0.93230 TO 0.93563	0.93146 TO 0.93647	791000
17	0.93404	+ OR - 0.00084	0.93320 TO 0.93487	0.93236 TO 0.93571	0.93153 TO 0.93654	786000
22	0.93411	+ OR - 0.00084	0.93327 TO 0.93495	0.93244 TO 0.93579	0.93160 TO 0.93663	781000
27	0.93414	+ OR - 0.00084	0.93330 TO 0.93498	0.93246 TO 0.93582	0.93161 TO 0.93666	776000
...						
772	0.93283	+ OR - 0.00390	0.92893 TO 0.93673	0.92503 TO 0.94062	0.92114 TO 0.94452	31000
777	0.93235	+ OR - 0.00435	0.92800 TO 0.93670	0.92364 TO 0.94106	0.91929 TO 0.94541	26000
782	0.93195	+ OR - 0.00515	0.92680 TO 0.93711	0.92165 TO 0.94226	0.91649 TO 0.94742	21000
787	0.92652	+ OR - 0.00555	0.92097 TO 0.93207	0.91542 TO 0.93762	0.90986 TO 0.94317	16000
792	0.92531	+ OR - 0.00693	0.91838 TO 0.93224	0.91145 TO 0.93917	0.90452 TO 0.94610	11000
797	0.92155	+ OR - 0.00885	0.91270 TO 0.93039	0.90385 TO 0.93924	0.89501 TO 0.94808	6000

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

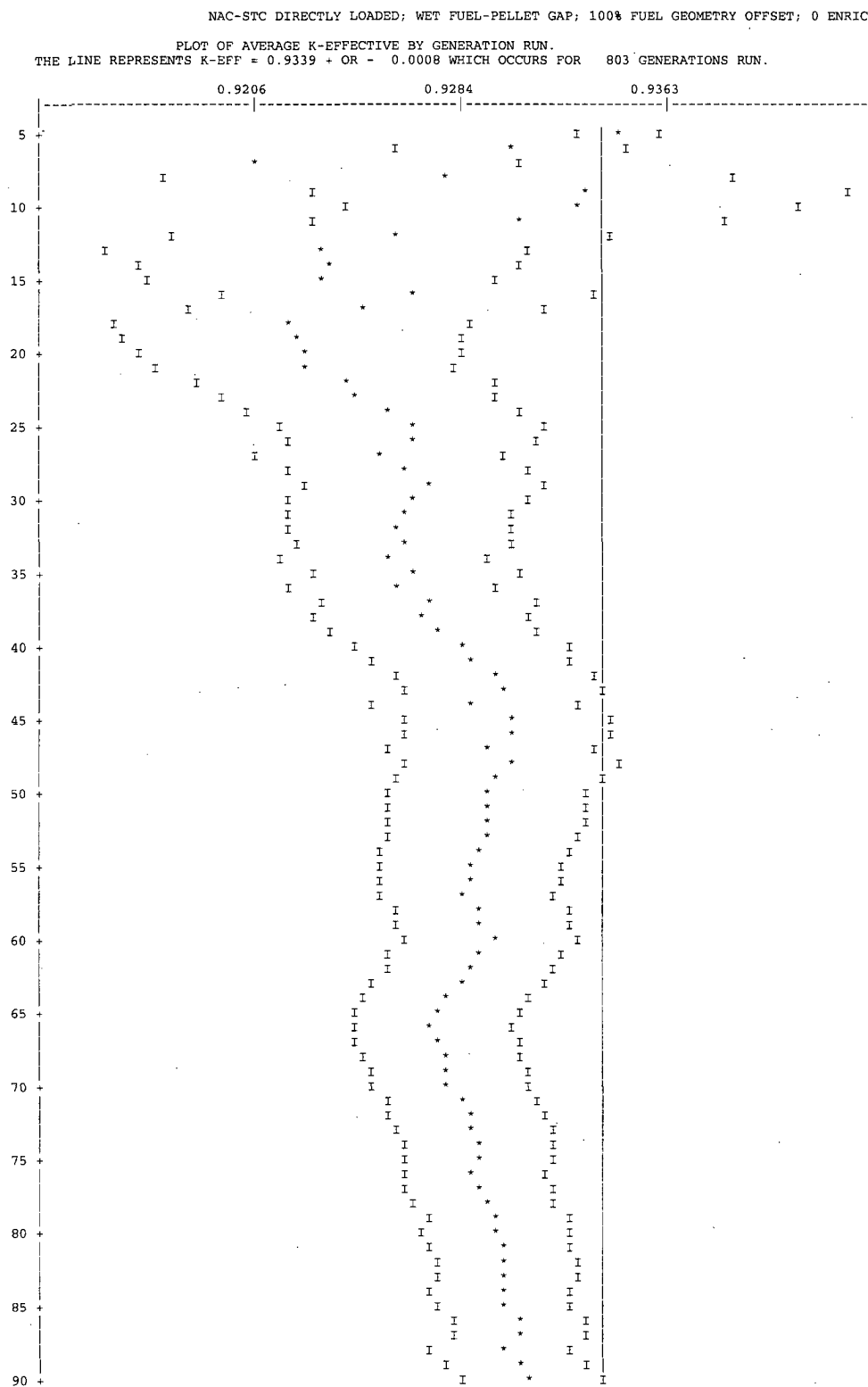


Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

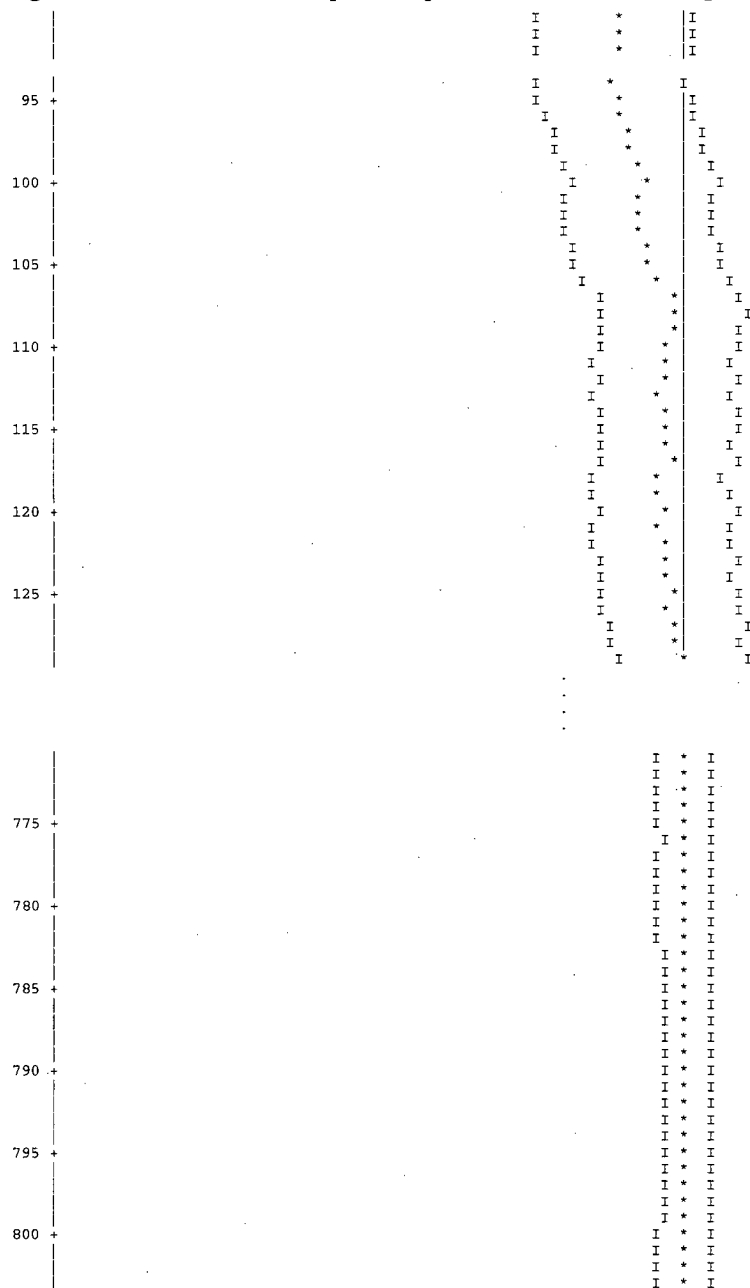


Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

```
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC  
FREQUENCY FOR GENERATIONS      4 TO 803  
0.8689 TO 0.8733    ***  
0.8733 TO 0.8776    ***  
0.8776 TO 0.8820    ****  
0.8820 TO 0.8864    *****  
0.8864 TO 0.8907    *****  
0.8907 TO 0.8951    *****  
0.8951 TO 0.8995    *****  
0.8995 TO 0.9038    *****  
0.9038 TO 0.9082    *****  
0.9082 TO 0.9125    *****  
0.9125 TO 0.9169    *****  
0.9169 TO 0.9213    *****  
0.9213 TO 0.9256    *****  
0.9256 TO 0.9300    *****  
0.9300 TO 0.9344    *****  
0.9344 TO 0.9387    *****  
0.9387 TO 0.9431    *****  
0.9431 TO 0.9475    *****  
0.9475 TO 0.9518    *****  
0.9518 TO 0.9562    *****  
0.9562 TO 0.9605    *****  
0.9605 TO 0.9649    *****  
0.9649 TO 0.9693    *****  
0.9693 TO 0.9736    *****  
0.9736 TO 0.9780    *****  
0.9780 TO 0.9824    *****  
0.9824 TO 0.9867    *****  
0.9867 TO 0.9911    *****  
0.9911 TO 0.9955    **  
0.9955 TO 0.9998    **  
0.9998 TO 1.0042    **  
1.0042 TO 1.0086    **
```

```
NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC  
FREQUENCY FOR GENERATIONS      204 TO 803  
0.8689 TO 0.8733    ***  
0.8733 TO 0.8776    *  
0.8776 TO 0.8820    ***  
0.8820 TO 0.8864    ****  
0.8864 TO 0.8907    *****  
0.8907 TO 0.8951    *****  
0.8951 TO 0.8995    *****  
0.8995 TO 0.9038    *****  
0.9038 TO 0.9082    *****  
0.9082 TO 0.9125    *****  
0.9125 TO 0.9169    *****  
0.9169 TO 0.9213    *****  
0.9213 TO 0.9256    *****  
0.9256 TO 0.9300    *****  
0.9300 TO 0.9344    *****  
0.9344 TO 0.9387    *****  
0.9387 TO 0.9431    *****  
0.9431 TO 0.9475    *****  
0.9475 TO 0.9518    *****  
0.9518 TO 0.9562    *****  
0.9562 TO 0.9605    *****  
0.9605 TO 0.9649    *****  
0.9649 TO 0.9693    *****  
0.9693 TO 0.9736    *****  
0.9736 TO 0.9780    *****  
0.9780 TO 0.9824    *****  
0.9824 TO 0.9867    *****  
0.9867 TO 0.9911    *****  
0.9911 TO 0.9955    **  
0.9955 TO 0.9998    **  
0.9998 TO 1.0042    **  
1.0042 TO 1.0086    **
```

Figure 6.7-10 CSAS25 Input/Output for Framatome-Cogema AFA Fuel (continued)

NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

FREQUENCY FOR GENERATIONS 404 TO 803

0.8689 TO 0.8733	***
0.8733 TO 0.8776	
0.8776 TO 0.8820	***
0.8820 TO 0.8864	***
0.8864 TO 0.8907	*****
0.8907 TO 0.8951	***
0.8951 TO 0.8995	***
0.8995 TO 0.9038	*****
0.9038 TO 0.9082	*****
0.9082 TO 0.9125	*****
0.9125 TO 0.9169	*****
0.9169 TO 0.9213	*****
0.9213 TO 0.9256	*****
0.9256 TO 0.9300	*****
0.9300 TO 0.9344	*****
0.9344 TO 0.9387	*****
0.9387 TO 0.9431	*****
0.9431 TO 0.9475	*****
0.9475 TO 0.9518	*****
0.9518 TO 0.9562	*****
0.9562 TO 0.9605	*****
0.9605 TO 0.9649	*****
0.9649 TO 0.9693	*****
0.9693 TO 0.9736	*****
0.9736 TO 0.9780	*****
0.9780 TO 0.9824	*****
0.9824 TO 0.9867	*****
0.9867 TO 0.9911	*****
0.9911 TO 0.9955	*
0.9955 TO 0.9998	*
0.9998 TO 1.0042	*
1.0042 TO 1.0086	**

NAC-STC DIRECTLY LOADED; WET FUEL-PELLET GAP; 100% FUEL GEOMETRY OFFSET; 0 ENRIC

FREQUENCY FOR GENERATIONS 604 TO 803

0.8689 TO 0.8733	*
0.8733 TO 0.8776	*
0.8776 TO 0.8820	*
0.8820 TO 0.8864	*
0.8864 TO 0.8907	*****
0.8907 TO 0.8951	*
0.8951 TO 0.8995	***
0.8995 TO 0.9038	*****
0.9038 TO 0.9082	*****
0.9082 TO 0.9125	*****
0.9125 TO 0.9169	*****
0.9169 TO 0.9213	*****
0.9213 TO 0.9256	*****
0.9256 TO 0.9300	*****
0.9300 TO 0.9344	*****
0.9344 TO 0.9387	*****
0.9387 TO 0.9431	*****
0.9431 TO 0.9475	*****
0.9475 TO 0.9518	*****
0.9518 TO 0.9562	*****
0.9562 TO 0.9605	*****
0.9605 TO 0.9649	*****
0.9649 TO 0.9693	*****
0.9693 TO 0.9736	*****
0.9736 TO 0.9780	*****
0.9780 TO 0.9824	*****
0.9824 TO 0.9867	*
0.9867 TO 0.9911	*
0.9911 TO 0.9955	*
0.9955 TO 0.9998	*
0.9998 TO 1.0042	*
1.0042 TO 1.0086	*

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions

```
PRIMARY MODULE ACCESS AND INPUT RECORD ( SCALE DRIVER - 95/03/29 - 09:06:37 )
MODULE CSAS25 WILL BE CALLED
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)
.
THIS IS A MODEL OF THE YNPS NAC-MPC BASKET
LOADED WITH 36 UNITED NUCLEAR TYPE A ASSEMBLIES
WITH MODIFIED TUBE ASSEMBLY 98
.
PRODUCED FOR THE YANKEE ROWE
STC LICENSE AMENDMENT
.
INTERIOR MODERATOR VOLUME FRACTION = 1.0
EXTERIOR MODERATOR VOLUME FRACTION = 1.0
CASK TO CASK PITCH = 300 cm
FLOODED PELLET CLAD GAP
NEUTRON SHIELD REMOVED
.
27GROUPNDF4 LATTICECELL
UO2      1      0.95      293.0 92235 4.0 92238 96.0 END
ZIRCALLOY 2      1.0      293.0      END
H2O      3      1.0      293.0      END
AL       4      1.0      293.0      END
SS304    5      1.0      293.0      END
B-10     6      DEN=2.6226 0.0450 293.0      END
B-11     6      DEN=2.6226 0.2736 293.0      END
C        6      DEN=2.6226 0.0927 293.0      END
AL       6      DEN=2.6226 0.5737 293.0      END
PB       7      1.0      293.0      END
H        8      DEN=1.6291 0.060 293.0      END
O        8      DEN=1.6291 0.425 293.0      END
C        8      DEN=1.6291 0.277 293.0      END
N        8      DEN=1.6291 0.020 293.0      END
AL       8      DEN=1.6291 0.214 293.0      END
B-10     8      DEN=1.6291 0.001 293.0      END
B-11     8      DEN=1.6291 0.004 293.0      END
H2O      9      1.0      293.0      END
H2O     10      1.0      293.0      END
END COMP
SQUAREPITCH 1.1887 0.7887 1 3 0.9271 2 0.8052 10 END
TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 cm) (IVF = 1.0) (EVF = 1.0)
READ PARAM RUN=YES PLT=YES GEN=1003 NPG=1000 TBA=15 RND=321 END PARAM
READ GEOM
.
WATER LEVEL UNIT CELLS
.
UNIT 1
COM='FUEL PIN CELL - BETWEEN DISKS'
CYLINDER 1 1 0.3943 2P2.1400
CYLINDER 10 1 0.4026 2P2.1400
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 2
COM='WATER CELL - BETWEEN DISKS'
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 3
COM='DISPLACEMENT CELL - BETWEEN DISKS'
CYLINDER 2 1 0.4635 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
UNIT 4
COM='INSTRUMENT TUBE CELL - BETWEEN DISKS'
CYLINDER 3 1 0.4998 2P2.1400
CYLINDER 5 1 0.5442 2P2.1400
CUBOID 3 1 4P0.5944 2P2.1400
.
DISK LEVEL UNIT CELLS (BOTH SS AND AL)
.
UNIT 5
COM='FUEL PIN CELL - WITH SS DISK'
CYLINDER 1 1 0.3943 2P0.6604
CYLINDER 10 1 0.4026 2P0.6604
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 6
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
COM='WATER CELL - WITH SS DISK'
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 7
COM='DISPLACEMENT CELL - WITH SS DISK'
CYLINDER 2 1 0.4635 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
UNIT 8
COM='INSTRUMENT TUBE CELL - WITH SS DISK'
CYLINDER 3 1 0.4998 2P0.6604
CYLINDER 5 1 0.5442 2P0.6604
CUBOID 3 1 4P0.5944 2P0.6604
'
WATER LEVEL BORAL SHEETS
'
UNIT 14
COM='X-X BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P9.144 2P0.0318 2P2.1400
CUBOID 4 1 2P9.144 2P0.0953 2P2.1400
UNIT 15
COM='Y-Y BORAL SHEET BETWEEN DISKS'
CUBOID 6 1 2P0.0318 2P9.144 2P2.1400
```


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
CUBOID 4 1 2P0.0953 2P9.144 2P2.1400
'
' DISK LEVEL BORAL SHEETS (AL AND SS)
'
UNIT 16
COM='X-X BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P9.144 2P0.0318 2P0.6604
CUBOID 4 1 2P9.144 2P0.0953 2P0.6604
UNIT 17
COM='Y-Y BORAL SHEET WITH SS DISK'
CUBOID 6 1 2P0.0318 2P9.144 2P0.6604
CUBOID 4 1 2P0.0953 2P9.144 2P0.6604
'
' WATER LEVEL WEB MATERIAL
'
UNIT 20
COM='WATER LEVEL WEB MATERIAL (SMALL) X-X'
CUBOID 3 1 2P10.4635 2P0.9716 2P2.1400
UNIT 21
COM='WATER LEVEL WEB MATERIAL (MEDIUM) X-X'
CUBOID 3 1 2P10.4635 2P1.0478 2P2.1400
UNIT 22
COM='WATER LEVEL WEB MATERIAL (LARGE) X-X'
CUBOID 3 1 2P10.4635 2P1.1208 2P2.1400
UNIT 23
COM='WATER LEVEL WEB MATERIAL (LONG) Y-Y'
CUBOID 3 1 2P1.1208 2P79.5249 2P2.1400
'
' SUPPORT DISK WEB MATERIAL
'
UNIT 30
COM='SUPPORT DISK WEB MATERIAL (SMALL) X-X'
CUBOID 5 1 2P10.4635 2P0.9716 2P0.6604
UNIT 31
COM='SUPPORT DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 5 1 2P10.4635 2P1.0478 2P0.6604
UNIT 32
COM='SUPPORT DISK WEB MATERIAL (LARGE) X-X'
CUBOID 5 1 2P10.4635 2P1.1208 2P0.6604
UNIT 33
COM='SUPPORT DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 5 1 2P1.1208 2P79.5249 2P0.6604
'
' HEAT TRANSFER DISK WEB MATERIAL
'
UNIT 40
COM='HEAT TRANSFER DISK WEB MATERIAL (SMALL) X-X'
CUBOID 4 1 2P10.4635 2P0.9716 2P0.6604
UNIT 41
COM='HEAT TRANSFER DISK WEB MATERIAL (MEDIUM) X-X'
CUBOID 4 1 2P10.4635 2P1.0478 2P0.6604
UNIT 42
COM='HEAT TRANSFER DISK WEB MATERIAL (LARGE) X-X'
CUBOID 4 1 2P10.4635 2P1.1208 2P0.6604
UNIT 43
COM='HEAT TRANSFER DISK WEB MATERIAL (LONG) Y-Y'
CUBOID 4 1 2P1.1208 2P79.5249 2P0.6604
'
' WATER LEVEL ASSEMBLY ARRAYS
'
UNIT 50
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL'
ARRAY 1 -9.0768 -9.0768 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400

UNIT 350
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL LR'
ARRAY 301 -9.9440 -9.0768 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400

UNIT 550
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL UR'
ARRAY 501 -9.9440 -9.9440 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400

UNIT 750
COM='FUEL TUBE AND ASSEMBLY - WATER LEVEL UL'
ARRAY 701 -9.0768 -9.9440 -2.1400
CUBOID 3 1 4P9.9441 2P2.1400
CUBOID 5 1 4P10.0661 2P2.1400
CUBOID 3 1 4P10.25681 2P2.1400
HOLE 14 0.0 10.1615 0.0
HOLE 14 0.0 -10.1615 0.0
HOLE 15 10.1615 0.0 0.0
HOLE 15 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P2.1400

UNIT 51
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 750 0.0 -0.1584 0.0
UNIT 52
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.0 0.1584 0.0
UNIT 53
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 550 -0.1584 0.0 0.0
UNIT 54
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 750 0.1584 0.0 0.0
UNIT 55
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 50 0.1584 0.1584 0.0
UNIT 56
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 350 -0.1584 0.1584 0.0
UNIT 57
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 750 0.1584 -0.1584 0.0
UNIT 58
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P2.1400
HOLE 550 -0.1584 -0.1584 0.0
UNIT 59
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P2.1400
```

DISK LEVEL ASSEMBLY ARRAYS

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
UNIT 60
COM='FUEL TUBE AND ASSEMBLY - DISK LEVEL'
ARRAY 2 -9.0768 -9.0768 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
```

```
UNIT 360
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL LR'
ARRAY 302 -9.9440 -9.0768 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
```

```
UNIT 560
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL UR'
ARRAY 502 -9.9440 -9.9440 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
```

```
UNIT 760
COM='FUEL TUBE AND ASSEMBLY - SUPPORT DISK LEVEL UL'
ARRAY 702 -9.0768 -9.9440 -0.6604
CUBOID 3 1 4P9.9441 2P0.6604
CUBOID 5 1 4P10.0661 2P0.6604
CUBOID 3 1 4P10.25681 2P0.6604
HOLE 16 0.0 10.1615 0.0
HOLE 16 0.0 -10.1615 0.0
HOLE 17 10.1615 0.0 0.0
HOLE 17 -10.1615 0.0 0.0
CUBOID 5 1 4P10.3051 2P0.6604
```

```
UNIT 61
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 760 0.0 -0.1584 0.0
```

```
UNIT 62
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.0 0.1584 0.0
```

```
UNIT 63
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 560 -0.1584 0.0 0.0
```

```
UNIT 64
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 760 0.1584 0.0 0.0
```

```
UNIT 65
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 60 0.1584 0.1584 0.0
```

```
UNIT 66
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 360 -0.1584 0.1584 0.0
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
UNIT 67
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 760 0.1584 -0.1584 0.0
UNIT 68
COM='ASSEMBLY CELL WITH 4 BORAL SHEETS -X -Y'
CUBOID 3 1 4P10.4635 2P0.6604
HOLE 560 -0.1584 -0.1584 0.0
UNIT 69
COM='CENTRAL HOLE'
CUBOID 3 1 4P10.4636 2P0.6604
'
' WATER LEVEL BASKET ARRAYS
UNIT 80
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 20 -10.4636 -33.6323 -2.1400
UNIT 81
COM='5X1 WATER LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 21 -10.4636 -33.6323 -2.1400
UNIT 82
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 22 -10.4636 -56.6549 -2.1400
UNIT 83
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 23 -10.4636 -56.6549 -2.1400
UNIT 84
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 24 -10.4636 -79.5251 -2.1400
UNIT 85
COM='13X1 WATER LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 25 -10.4636 -79.5251 -2.1400
UNIT 86
COM='13X1 WATER LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 26 -10.4636 -79.5251 -2.1400
'
' SUPPORT DISK LEVEL BASKET ARRAYS
UNIT 90
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 30 -10.4636 -33.6323 -0.6604
UNIT 91
COM='5X1 SUPPORT DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 31 -10.4636 -33.6323 -0.6604
UNIT 92
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 32 -10.4636 -56.6549 -0.6604
UNIT 93
COM='9X1 WATER LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 33 -10.4636 -56.6549 -0.6604
UNIT 94
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 34 -10.4636 -79.5251 -0.6604
UNIT 95
COM='13X1 SUPPORT DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 35 -10.4636 -79.5251 -0.6604
UNIT 96
COM='13X1 SUPPORT DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 36 -10.4636 -79.5251 -0.6604
'
' HEAT TRANSFER DISK LEVEL BASKET ARRAYS
UNIT 100
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY -X)'
ARRAY 40 -10.4636 -33.6323 -0.6604
UNIT 101
COM='5X1 HEAT TRANSFER DISK LEVEL ARRAY (SMALL ARRAY X)'
ARRAY 41 -10.4636 -33.6323 -0.6604
UNIT 102
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY -X)'
ARRAY 42 -10.4636 -56.6549 -0.6604
UNIT 103
COM='9X1 HEAT TRANSFER DISK LEVEL ARRAY (MEDIUM ARRAY X)'
ARRAY 43 -10.4636 -56.6549 -0.6604
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
UNIT 104
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY -X)'
ARRAY 44 -10.4636 -79.5251 -0.6604
UNIT 105
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (MIDDLE LARGE ARRAY)'
ARRAY 45 -10.4636 -79.5251 -0.6604
UNIT 106
COM='13X1 HEAT TRANSFER DISK LEVEL ARRAY (LARGE ARRAY X)'
ARRAY 46 -10.4636 -79.5251 -0.6604
,
BASKET ARRAY IN TRANSPORT CASK OVERPACK (LEVEL CONSTRUCTION)
,
UNIT 110
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - WATER LEVEL'
ARRAY 50 -33.6323 -79.5251 -2.1400
CYLINDER 3 1 88.1253 2P2.1400
HOLE 80 -69.0614 0.0 0.0
HOLE 82 -46.1912 0.0 0.0
HOLE 81 69.0614 0.0 0.0
HOLE 83 46.1912 0.0 0.0
CYLINDER 5 1 89.7128 2P2.1400
CYLINDER 3 1 90.170 2P2.1400
CYLINDER 5 1 93.98 2P2.1400
CYLINDER 7 1 103.4288 2P2.1400
CYLINDER 5 1 110.109 2P2.1400
CYLINDER 9 1 124.714 2P2.1400
CYLINDER 9 1 125.349 2P2.1400
CUBOID 9 1 4P150 2P2.1400
UNIT 111
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - SUPPORT DISK LEVEL'
ARRAY 51 -33.6323 -79.5251 -0.6604
CYLINDER 5 1 87.6046 2P0.6604
HOLE 90 -69.0614 0.0 0.0
HOLE 92 -46.1912 0.0 0.0
HOLE 91 69.0614 0.0 0.0
HOLE 93 46.1912 0.0 0.0
CYLINDER 3 1 88.1253 2P0.6604
CYLINDER 5 1 89.7128 2P0.6604
CYLINDER 3 1 90.170 2P0.6604
CYLINDER 5 1 93.98 2P0.6604
CYLINDER 7 1 103.4288 2P0.6604
CYLINDER 5 1 110.109 2P0.6604
CYLINDER 9 1 124.714 2P0.6604
CYLINDER 9 1 125.349 2P0.6604
CUBOID 9 1 4P150 2P0.6604
UNIT 112
COM='BASKET ARRAY IN TRANSPORT CASK OVERPACK - HEAT TRANSFER DISK LEVEL'
ARRAY 52 -33.6323 -79.5251 -0.6604
CYLINDER 4 1 87.2490 2P0.6604
HOLE 100 -69.0614 0.0 0.0
HOLE 102 -46.1912 0.0 0.0
HOLE 101 69.0614 0.0 0.0
HOLE 103 46.1912 0.0 0.0
CYLINDER 3 1 88.1253 2P0.6604
CYLINDER 5 1 89.7128 2P0.6604
CYLINDER 3 1 90.170 2P0.6604
CYLINDER 5 1 93.98 2P0.6604
CYLINDER 7 1 103.4288 2P0.6604
CYLINDER 5 1 110.109 2P0.6604
CYLINDER 9 1 124.714 2P0.6604
CYLINDER 9 1 125.349 2P0.6604
CUBOID 9 1 4P150 2P0.6604
,
UNIT 130
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- WATER LEVEL'
ARRAY 3 -8.8736 -8.8736 -2.1400
CUBOID 3 1 4P10.1473 2P2.1400
CUBOID 5 1 4P10.26922 2P2.1400
CUBOID 3 1 4P10.3051 2P2.1400
,
UNIT 330
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- WATER LEVEL LR'
ARRAY 303 -10.1472 -8.8736 -2.1400
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
CUBOID 3 1 4P10.1473          2P2.1400
CUBOID 5 1 4P10.26922         2P2.1400
CUBOID 3 1 4P10.3051          2P2.1400

UNIT 530
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- WATER LEVEL UR'
ARRAY 503 -10.1472 -10.1472 -2.1400
CUBOID 3 1 4P10.1473          2P2.1400
CUBOID 5 1 4P10.26922         2P2.1400
CUBOID 3 1 4P10.3051          2P2.1400

UNIT 730
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- WATER LEVEL UL'
ARRAY 703 -8.8736 -10.1472 -2.1400
CUBOID 3 1 4P10.1473          2P2.1400
CUBOID 5 1 4P10.26922         2P2.1400
CUBOID 3 1 4P10.3051          2P2.1400

UNIT 131
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL +X +Y'
CUBOID 3 1 4P10.4635          2P2.1400
HOLE 130 0.1584 0.1584 0.0
UNIT 132
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL -X +Y'
CUBOID 3 1 4P10.4635          2P2.1400
HOLE 330 -0.1584 0.1584 0.0
UNIT 133
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL +X -Y'
CUBOID 3 1 4P10.4635          2P2.1400
HOLE 730 0.1584 -0.1584 0.0
UNIT 134
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - WATER LEVEL -X -Y'
CUBOID 3 1 4P10.4635          2P2.1400
HOLE 530 -0.1584 -0.1584 0.0

UNIT 140
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- DISK LEVEL'
ARRAY 4 -8.8736 -8.8736 -0.6604
CUBOID 3 1 4P10.1473          2P0.6604
CUBOID 5 1 4P10.26922         2P0.6604
CUBOID 3 1 4P10.3051          2P0.6604

UNIT 340
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- DISK LEVEL LR'
ARRAY 304 -10.1472 -8.8736 -0.6604
CUBOID 3 1 4P10.1473          2P0.6604
CUBOID 5 1 4P10.26922         2P0.6604
CUBOID 3 1 4P10.3051          2P0.6604

UNIT 540
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- DISK LEVEL UR'
ARRAY 504 -10.1472 -10.1472 -0.6604
CUBOID 3 1 4P10.1473          2P0.6604
CUBOID 5 1 4P10.26922         2P0.6604
CUBOID 3 1 4P10.3051          2P0.6604

UNIT 740
COM='OVERSIZED FUEL TUBE AND ASSEMBLY 98- DISK LEVEL UL'
ARRAY 704 -8.8736 -10.1472 -0.6604
CUBOID 3 1 4P10.1473          2P0.6604
CUBOID 5 1 4P10.26922         2P0.6604
CUBOID 3 1 4P10.3051          2P0.6604

UNIT 141
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - DISK LEVEL +X +Y'
CUBOID 3 1 4P10.4635          2P0.6604
HOLE 140 0.1584 0.1584 0.0
UNIT 142
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - DISK LEVEL -X +Y'
CUBOID 3 1 4P10.4635          2P0.6604
HOLE 340 -0.1584 0.1584 0.0
```

```
UNIT 143  
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - DISK LEVEL +X -Y'  
CUBOID 3 1 4P10.4635 2P0.6604  
HOLE 740 0.1584 -0.1584 0.0  
  
UNIT 144  
COM='OVERSIZED FUEL TUBE AND ASSEMBLY - DISK LEVEL -X -Y'  
CUBOID 3 1 4P10.4635 2P0.6604  
HOLE 540 0.1584 -0.1584 0.0
```

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```

END FILL
ARA=304 NUX=16 NUY=16 NUZ=1 FILL
6 6 6 6 6 6 6 6 7 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 6 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 6 5 5 5 5 5 5 5 5 5

```


[illegible][illegible][illegible]

END	FILL
ARA=503	NUX=16 NUY=16 NUZ=1 FILL
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 2	
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 2	
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 2	
1 1 1 1 1 1 1 2 1 1 1 1 1 1 1 2	
1 1 1 1 1 1 2 2 1 1 1 1 1 1 1 2	
1 1 1 1 1 2 1 1 1 2 1 1 1 1 1 2	
1 1 1 1 2 1 2 1 2 1 2 1 1 1 1 2	
1 1 1 2 1 1 1 4 1 1 1 2 1 1 1 2	
1 1 1 2 1 2 1 2 1 2 1 1 1 1 1 3	
1 1 1 1 1 2 1 1 1 2 1 1 1 1 1 1	
1 1 1 1 1 1 2 1 2 1 1 1 1 1 1 1	
1 1 1 1 1 1 1 1 2 1 1 1 1 1 1 1	
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	
2 2 2 2 2 2 2 2 3 1 1 1 1 1 1 1	

[illegible]

6.7-300

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
5 5 5 5 5 5 5 6 5 6 5 5 5 5 5 5
5 5 5 5 5 5 5 6 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5
5 5 5 5 5 5 5 7 6 6 6 6 6 6 6 6
END FILL
```

WATER LEVEL ARRAYS

ARA=20 NUX=1 NUY=5 NUZ=1
FILL

57
22
57
22
55

END FILL

ARA=21 NUX=1 NUY=5 NUZ=1

FILL

58
22
56
22
56

END FILL

ARA=22 NUX=1 NUY=9 NUZ=1

FILL

131
21
57
22
57
22
55

21

133

END FILL

ARA=23 NUX=1 NUY=9 NUZ=1

FILL

132
21
58
22
56
22
56
21

134

END FILL

ARA=24 NUX=1 NUY=13 NUZ=1

FILL

55
20
55
21
57
22

55

22

55

21

57

20

57

END FILL

ARA=25 NUX=1 NUY=13 NUZ=1

FILL

56
20
56
21
58
22

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
59
22
55
21
57
20
57
END FILL
ARA=26 NUX=1 NUY=13 NUZ=1
FILL
56
20
56
21
58
22
58
22
56
21
58
20
58
END FILL
SUPPORT DISK LEVEL ARRAYS
ARA=30 NUX=1 NUY=5 NUZ=1
FILL
67
32
67
32
65
END FILL
ARA=31 NUX=1 NUY=5 NUZ=1
FILL
68
32
66
32
66
END FILL
ARA=32 NUX=1 NUY=9 NUZ=1
FILL
141
31
67
32
67
32
65
31
143
END FILL
ARA=33 NUX=1 NUY=9 NUZ=1
FILL
142
31
68
32
66
32
66
31
144
END FILL
ARA=34 NUX=1 NUY=13 NUZ=1
FILL
65
30
65
31
67
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
32
65
32
65
31
67
30
67
END FILL
ARA=35 NUX=1 NUY=13 NUZ=1
FILL
66
30
66
31
68
32
69
32
65
31
67
30
67
END FILL
ARA=36 NUX=1 NUY=13 NUZ=1
FILL
66
30
66
31
68
32
68
32
66
31
68
30
68
END FILL
HEAT TRANSFER DISK LEVEL ARRAYS
ARA=40 NUX=1 NUY=5 NUZ=1
FILL
67
42
67
42
65
END FILL
ARA=41 NUX=1 NUY=5 NUZ=1
FILL
68
42
66
42
66
END FILL
ARA=42 NUX=1 NUY=9 NUZ=1
FILL
141
41
67
42
67
42
65
41
143
END FILL
ARA=43 NUX=1 NUY=9 NUZ=1
FILL
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
142
41
68
42
66
42
66
41
144
END FILL
ARA=44 NUX=1 NUY=13 NUZ=1
FILL
65
40
65
41
67
42
65
42
65
41
67
40
67
END FILL
ARA=45 NUX=1 NUY=13 NUZ=1
FILL
66
40
66
41
68
42
69
42
65
41
67
40
67
END FILL
ARA=46 NUX=1 NUY=13 NUZ=1
FILL
66
40
66
41
68
42
68
42
66
41
68
40
68
END FILL
MAJOR ARRAYS
ARA=50 NUX=5 NUY=1 NUZ=1
FILL
86 23 85 23 84
END FILL
ARA=51 NUX=5 NUY=1 NUZ=1
FILL
96 33 95 33 94
END FILL
ARA=52 NUX=5 NUY=1 NUZ=1
FILL
106 43 105 43 104
END FILL
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
GLOBAL ARRAY
ARA=60 NUX=1 NUZ=1 NUZ=4
FILL
112
110
111
110
END FILL
END ARRAY
READ BOUNDS ZFC=PER YXF=MIR END BOUNDS
READ PLOT
SCR=YES PIC=MAT LPI=10
UAX=1.0 VDN=-1.0 NAX=1500

WHOLE BASKET HORIZONTAL SLICES

TTL='BASKET X-Y CROSS SECTION AT Z= 0.635 HEAT TRANSFER DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 0.635
XLR= 130 YLR= -130 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 3.44 WATER LEVEL'
XUL= -130 YUL= 130 ZUL= 3.44
XLR= 130 YLR= -130 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y CROSS SECTION AT Z= 6.236 SS DISK LEVEL'
XUL= -130 YUL= 130 ZUL= 6.236
XLR= 130 YLR= -130 ZLR= 6.236
UAX=1.0 VDN=-1.0 NAX=1500 END

HEAT TRANSFER DISK LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I HEAT TRANSFER DISK'
XUL= 12.0 YUL= 80 ZUL= 0.635
XLR= 80.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II HEAT TRANSFER DISK'
XUL= 12.0 YUL= -12.0 ZUL= 0.635
XLR= 80 YLR= -80 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III HEAT TRANSFER DISK'
XUL= -80.0 YUL= -12.0 ZUL= 0.635
XLR= -12.0 YLR= -80.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV HEAT TRANSFER DISK'
XUL= -80.0 YUL= 80.0 ZUL= 0.635
XLR= -12.0 YLR= 12.0 ZLR= 0.635
UAX=1.0 VDN=-1.0 NAX=1500 END

WATER LEVEL BASKET QUADRANTS

TTL='BASKET X-Y QUADRANT I WATER LEVEL'
XUL= 12.0 YUL= 80 ZUL= 3.44
XLR= 80.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT II WATER LEVEL'
XUL= 12.0 YUL= -12.0 ZUL= 3.44
XLR= 80 YLR= -80 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT III WATER LEVEL'
XUL= -80.0 YUL= -12.0 ZUL= 3.44
XLR= -12.0 YLR= -80.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END
TTL='BASKET X-Y QUADRANT IV WATER LEVEL'
XUL= -80.0 YUL= 80.0 ZUL= 3.44
XLR= -12.0 YLR= 12.0 ZLR= 3.44
UAX=1.0 VDN=-1.0 NAX=1500 END

VERTICAL SLICES

TTL='BASKET X-Z CROSS SECTION ALUMINUM LEVEL (MIDDLE OF FUEL PIN)'
XUL= -90 YUL=0.4 ZUL= 1.27
XLR= 90 YLR=0.4 ZLR= -.1
UAX=1.0 VDN=-1.0 NAX=1500 END
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
TTL='BASKET X-Z CROSS SECTION WATER LEVEL (MIDDLE OF FUEL PIN)'  
XUL= -90 YUL=0.4 ZUL= 4.318  
XLR= 90 YLR=0.4 ZLR= 1.27  
UAX=1.0 WDN=-1.0 NAX=1500 END  
TTL='BASKET X-Z CROSS SECTION SS LEVEL (MIDDLE OF FUEL PIN)'  
XUL= -90 YUL=0.4 ZUL= 6.858  
XLR= 90 YLR=0.4 ZLR= 5.588  
UAX=1.0 WDN=-1.0 NAX=1500 END  
TTL='BASKET X-Z CROSS SECTION ENTIRE MODEL (MIDDLE OF FUEL PIN)'  
XUL= -90 YUL=0.4 ZUL= 12  
XLR= 90 YLR=0.4 ZLR= 0  
UAX=1.0 WDN=-1.0 NAX=1500 END  
END PLOT  
END DATA  
  
SECONDARY MODULE 000008 HAS BEEN CALLED.  
  
MODULE 000008 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 0.44 (SECONDS).  
  
SECONDARY MODULE 000002 HAS BEEN CALLED.  
  
MODULE 000002 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 3.07 (SECONDS).  
  
SECONDARY MODULE 000009 HAS BEEN CALLED.  
  
MODULE 000009 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 697.72 (SECONDS).  
  
MODULE CSAS25 IS FINISHED. COMPLETION CODE 0. CPU TIME USED 703.54 (SECONDS).
```


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

CCCCCCCCC	SSSSSSSSS	AAAAAAA	SSSSSSSSS	222222222	55555555555
CCCCCCCCC	SSSSSSSSS	AAAAAAA	SSSSSSSSS	22222222222	55555555555
CC CC	SS SS	AA AA	SS SS	22 22	55
CC	SS	AA AA	SS	22	55
CC	SS	AA AA	SS	22	55
CC	SSSSSSSSS	AAAAAAA	SSSSSSSSS	22	55555555555
CC	SSSSSSSSS	AAAAAAA	SSSSSSSSS	22	55555555555
CC	SS	AA AA	SS	22	55
CC	SS	AA AA	SS	22	55
CC CC	SS SS	AA AA	SS SS	22	55 55
CCCCCCCCC	SSSSSSSSS	AA AA	SSSSSSSSS	22222222222	55555555555
CCCCCCCCC	SSSSSSSSS	AA AA	SSSSSSSSS	22222222222	55555555555

SSSSSSSSS	CCCCCCCCC	AAAAAAA	LL	EEEEEEEEEE	PPPPPPPPPP	CCCCCCCCC
SSSSSSSSS	CCCCCCCCC	AAAAAAA	LL	EEEEEEEEEE	PPPPPPPPPP	CCCCCCCCC
SS SS	CC CC	AA AA	LL	EE	PP PP	CC CC
SS	CC	AA AA	LL	EE	PP PP	CC
SS	CC	AA AA	LL	EE	PP PP	CC
SSSSSSSSS	CC	AAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SSSSSSSSS	CC	AAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SS	CC	AA AA	LL	EE	PP	CC
SS	CC	AA AA	LL	EE	PP	CC
SS	CC	AA AA	LL	EE	PP	CC
SSSSSSSSS	CCCCCCCCC	AA AA	LLLLLLLLLLLL	EEEEEEEEEE	PP	CCCCCCCCC
SSSSSSSSS	CCCCCCCCC	AA AA	LLLLLLLLLLLL	EEEEEEEEEE	PP	CCCCCCCCC

0000000	333333333	//	222222222	55555555555	//	0000000	222222222
000000000	33333333333	//	22222222222	55555555555	//	000000000	22222222222
00 00	33 33	//	22 22	55	//	00 00	22 22
00 00	33	//	22	55	//	00 00	22
00 00	33	//	22	55	//	00 00	22
00 00	333	//	22	55555555555	//	00 00	22
00 00	333	//	22	55555555555	//	00 00	22
00 00	33	//	22	55	//	00 00	22
00 00	33	//	22	55	//	00 00	22
00 00	33 33	//	22	55 55	//	00 00	22
000000000	33333333333	//	22222222222	55555555555	//	000000000	22222222222
0000000	333333333	//	22222222222	55555555555	//	0000000	22222222222

11	333333333	0000000	11	222222222	222222222
111	33333333333	000000000	111	22222222222	22222222222
1111	33 33	00 00	1111	22 22	22 22
11	33	00 00	11	22	22
11	33	00 00	11	22	22
11	333	00 00	11	22	22
11	333	00 00	11	22	22
11	33	00 00	11	22	22
11	33	00 00	11	22	22
11	33	00 00	11	22	22
11111111	33333333333	000000000	11111111	22222222222	22222222222
11111111	333333333	0000000	11111111	22222222222	22222222222

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

**** PROBLEM PARAMETERS ****

LIB 27GROUPNDF4 LIBRARY
MX 10 MIXTURES
MSC 19 COMPOSITION SPECIFICATIONS
IZM 4 MATERIAL ZONES
GE LATTICECELL GEOMETRY
MORE 0 0/1 DO NOT READ/READ OPTIONAL PARAMETER DATA
MSLN 0 FUEL SOLUTIONS

**** PROBLEM COMPOSITION DESCRIPTION ****

SC UO2 STANDARD COMPOSITION
MX 1 MIXTURE NO.
VF 0.9500 VOLUME FRACTION
ROTH 10.9600 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
92000 1.00 ATOM/MOLECULE
92235 4.000 WT%
92238 96.000 WT%
8016 2.00 ATOMS/MOLECULE

END

SC ZIRCALLOY STANDARD COMPOSITION
MX 2 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 6.5600 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
40302 1.00 ATOM/MOLECULE

END

SC H2O STANDARD COMPOSITION
MX 3 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 0.9982 THEORETICAL DENSITY
NEL 2 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
1001 2.00 ATOMS/MOLECULE
8016 1.00 ATOM/MOLECULE

END

SC AL STANDARD COMPOSITION
MX 4 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 2.7020 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
13027 1.00 ATOM/MOLECULE

END

SC SS304 STANDARD COMPOSITION
MX 5 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 7.9200 THEORETICAL DENSITY
NEL 4 NO. ELEMENTS
ICP 0 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
24304 19.000 WT%
25055 2.000 WT%
26304 69.500 WT%
28304 9.500 WT%

END

SC B-10 STANDARD COMPOSITION
MX 6 MIXTURE NO.
VF 0.0450 VOLUME FRACTION
ROTH 2.6226 SPECIFIED DENSITY

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
5010 1.00 ATOM/MOLECULE
END

SC B-11 STANDARD COMPOSITION
MX 6 MIXTURE NO.
VF 0.2736 VOLUME FRACTION
ROTH 2.6226 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
5011 1.00 ATOM/MOLECULE
END

SC C STANDARD COMPOSITION
MX 6 MIXTURE NO.
VF 0.0927 VOLUME FRACTION
ROTH 2.6226 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
6012 1.00 ATOM/MOLECULE
END

SC AL STANDARD COMPOSITION
MX 6 MIXTURE NO.
VF 0.5737 VOLUME FRACTION
ROTH 2.6226 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
13027 1.00 ATOM/MOLECULE
END

SC PB STANDARD COMPOSITION
MX 7 MIXTURE NO.
VF 1.0000 VOLUME FRACTION
ROTH 11.3440 THEORETICAL DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
82000 1.00 ATOM/MOLECULE
END

SC H STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.0600 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
1001 1.00 ATOM/MOLECULE
END

SC O STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.4250 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
8016 1.00 ATOM/MOLECULE
END

SC C STANDARD COMPOSITION
MX 8 MIXTURE NO.
VF 0.2770 VOLUME FRACTION
ROTH 1.6291 SPECIFIED DENSITY
NEL 1 NO. ELEMENTS
ICP 1 0/1 MIXTURE/COMPOUND
TEMP 293.0 DEG KELVIN
6012 1.00 ATOM/MOLECULE

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
END

SC N          STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.0200 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              7014      1.00 ATOM/MOLECULE
END

SC AL         STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.2140 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              13027     1.00 ATOM/MOLECULE
END

SC B-10       STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.0010 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              5010      1.00 ATOM/MOLECULE
END

SC B-11       STANDARD COMPOSITION
MX            8 MIXTURE NO.
VF            0.0040 VOLUME FRACTION
ROTH          1.6291 SPECIFIED DENSITY
NEL           1 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              5011      1.00 ATOM/MOLECULE
END

SC H2O        STANDARD COMPOSITION
MX            9 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          0.9982 THEORETICAL DENSITY
NEL           2 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              1001      2.00 ATOMS/MOLECULE
              8016      1.00 ATOM/MOLECULE
END

SC H2O        STANDARD COMPOSITION
MX            10 MIXTURE NO.
VF            1.0000 VOLUME FRACTION
ROTH          0.9982 THEORETICAL DENSITY
NEL           2 NO. ELEMENTS
ICP           1 0/1 MIXTURE/COMPOUND
TEMP          293.0 DEG KELVIN
              1001      2.00 ATOMS/MOLECULE
              8016      1.00 ATOM/MOLECULE
END

**** PROBLEM GEOMETRY ****

CTP SQUAREPITCH CELL TYPE
PITCH         1.1887 CM CENTER TO CENTER SPACING
FUELOD        0.7887 CM FUEL DIAMETER OR SLAB THICKNESS
MFUEL         1 MIXTURE NO. OF FUEL
MMOD          3 MIXTURE NO. OF MODERATOR
CLADOD        0.9271 CM CLAD OUTER DIAMETER
MCLAD         2 MIXTURE NO. OF CLAD
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal
Conditions (continued)

GAPOD 0.8052 CM GAP OUTER DIAMETER
MGAP 10 MIXTURE NO. OF GAP

ZONE SPECIFICATIONS FOR LATTICECELL GEOMETRY

ZONE 1 IS FUEL
ZONE 2 IS GAP
ZONE 3 IS CLAD
ZONE 4 IS MOD

```

***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)*
***
***** DATA LIBRARY INFORMATION *****
UNIT      VOLUME
NUMBER    NAME      UNIT FUNCTION
-----
89      M:\scale43\DATALIB\FT89F001      STANDARD COMPOSITION LIBRARY
82      M:\scale43\DATALIB\FT82F001      CROSS SECTION LIBRARY
11      D:\Projects\YAEC\5501\Runs\reflected_disk_st      SHORT CROSS SECTION LIBRARY
90      D:\Projects\YAEC\5501\Runs\reflected_disk_st      INPUT DATA DIRECT ACCESS

STANDARD COMPOSITION LIBRARY DATA
-----
UNIT NUMBER : 89
DATASET NAME : M:\scale43\DATALIB\FT89F001
LIBRARY TITLE: SCALE-4 STANDARD COMPOSITION LIBRARY
                637 STANDARD COMPOSITIONS, 490 NUCLIDES
                90 ELEMENTS WITH VARIABLE ISOTOPIC DISTRIBUTIONS.
CREATION DATE: 6/30/95

CROSS SECTION LIBRARY DATA
-----
UNIT NUMBER : 82
DATASET NAME : M:\scale43\DATALIB\FT82F001
LIBRARY TITLE: SCALE 4.2 - 27 GROUP NEUTRON GROUP LIBRARY
                BASED ON ENDF-B VERSION 4 DATA
                COMPILED FOR NRC           1/27/89
                LAST UPDATED
                L.M.PETRIE   -   ORNL
                                08/12/94

..... 0 IO'S WERE USED BEFORE READING KENO V DATA .....
..... 0 IO'S WERE USED READING THE KENO V PARAMETER DATA .....
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

KK	KK	EEEEEEEEEEEE	NN	NN	0000000000	VV	VV
KK	KK	EEEEEEEEEEEE	NNN	NN	000000000000	VV	VV
KK	KK	EE	NNNN	NN	OO	OO	VV
KK	KK	EE	NN NN	NN	OO	OO	VV
KK	KK	EE	NN NN	NN	OO	OO	VV
KKKKKKK		EEEEEEEE	NN NN	NN	OO	OO	VV
KKKKKKK		EEEEEEEE	NN NN	NN	OO	OO	VV
KK	KK	EE	NN NN	NN	OO	OO	VV
KK	KK	EE	NN NN	NN	OO	OO	VV
KK	KK	EE	NN	NNN	OO	OO	VV
KK	KK	EEEEEEEEEEEE	NN	NNN	000000000000	VVV	V
KK	KK	EEEEEEEEEEEE	NN	NN	0000000000		

SSSSSSSSSS	CCCCCCCCCC	AAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SSSSSSSSSS	CCCCCCCCCC	AAAAAAA	LL	EEEEEEEEEEEE	PPPPPPPPPP	CCCCCCCCCC
SS	SS	CC	CC	AA	AA	CC
SS	CC	AA	AA	LL	EE	CC
SS	CC	AA	AA	LL	EE	CC
SSSSSSSSSS	CC	AAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SSSSSSSSSS	CC	AAAAAAA	LL	EEEEEEEE	PPPPPPPPPP	CC
SS	CC	AA	AA	LL	EE	CC
SS	CC	AA	AA	LL	EE	CC
SS	CC	AA	AA	LL	EE	CC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	EEEEEEEEEEEE	CCCCCCCCCC
SSSSSSSSSS	CCCCCCCCCC	AA	AA	LLLLLLLLLLLL	EEEEEEEEEEEE	CCCCCCCCCC

0000000	3333333333	//	2222222222	5555555555	//	0000000	2222222222
000000000	333333333333	//	222222222222	555555555555	//	000000000	222222222222
00	00	33	33	55	55	00	00
00	00	33	33	55	55	00	00
00	00	33	33	55	55	00	00
00	00	333	33	5555555555	55	00	00
00	00	333	33	555555555555	55	00	00
00	00	33	33	55	55	00	00
00	00	33	33	55	55	00	00
00	00	33	33	55	55	00	00
00	00	33	33	55	55	00	00
000000000	333333333333	//	222222222222	555555555555	//	000000000	222222222222
0000000	33333333333	//	222222222222	55555555555	//	0000000	222222222222

11	3333333333		0000000	11	2222222222	7777777777
111	333333333333		000000000	111	222222222222	77777777777
1111	33	33	00	00	22	77
11	33	33	00	00	22	77
11	33	33	00	00	22	77
11	333	33	00	00	22	77
11	333	33	00	00	22	77
11	33	33	00	00	22	77
11	33	33	00	00	22	77
11	33	33	00	00	22	77
1111111	333333333333		000000000	1111111	222222222222	77
1111111	33333333333		0000000	1111111	222222222222	77

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

SSSSSSSSSS	CCCCCCCCCC	AAAAAAAA	LL	EEEEEEEEEEEE		PPPPPPPPPP	CCCCCCCCCC				
SSSSSSSSSSSS	CCCCCCCCCCCC	AAAAAAAAAA	LL	EEEEEEEEEEEE		PPPPPPPPPPPP	CCCCCCCCCCCC				
SS	SS	CC	CC	AA	AA	LL	EE	PP	PP	CC	CC
SS		CC		AA	AA	LL	EE		PP	PP	CC
SS		CC		AA	AA	LL	EE		PP	PP	CC
SSSSSSSSSSSS	CC			AAAAAAAAAAAA	LL		EEEEEEEE	-----	PPPPPPPPPPPP	CC	
SSSSSSSSSSSS	CC			AAAAAAAAAAAA	LL		EEEEEEEE	-----	PPPPPPPPPPPP	CC	
	SS	CC		AA	AA	LL	EE		PP		CC
	SS	CC		AA	AA	LL	EE		PP		CC
SS	SS	CC	CC	AA	AA	LL	EE		PP		CC
SSSSSSSSSSSS	CCCCCCCCCCCC	AA	AA	LLLLLLLLLLLL		EEEEEEEEEEEE			PP		CCCCCCCCCCCC
SSSSSSSSSSSS	CCCCCCCCCCCC	AA	AA	LLLLLLLLLLLL		EEEEEEEEEEEE			PP		CCCCCCCCCCCC

```
*****  
*****  
***** PROGRAM VERIFICATION INFORMATION *****  
*****  
***** CODE SYSTEM: SCALE-PC VERSION: 4.3 *****  
*****  
*****  
*****  
*****  
***** PROGRAM: O0O009 *****  
*****  
***** CREATION DATE: 03/08/96 *****  
*****  
***** VOLUME: Eng *****  
*****  
***** LIBRARY: M:\SCALE43\WIN_NT\EXE *****  
*****  
*****  
***** PRODUCTION CODE: KENOVA *****  
*****  
***** VERSION: 3.1 *****  
*****  
***** JOBNAME: SCALE-PC *****  
*****  
***** DATE OF EXECUTION: 03/25/02 *****  
*****  
***** TIME OF EXECUTION: 13:01:27 *****  
*****  
*****  
*****
```


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```

*****
***
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)***
***
*****
***          NUMERIC PARAMETERS          ***
***
***          TME          MAXIMUM PROBLEM TIME (MIN)          30.00          ***
***
***          TBA          TIME PER GENERATION (MIN)          15.00          ***
***
***          GEN          NUMBER OF GENERATIONS          1003          ***
***
***          NPG          NUMBER PER GENERATION          1000          ***
***
***          NSK          NUMBER OF GENERATIONS TO BE SKIPPED          3          ***
***
***          BEG          BEGINNING GENERATION NUMBER          1          ***
***
***          RES          GENERATIONS BETWEEN CHECKPOINTS          0          ***
***
***          X1D          NUMBER OF EXTRA 1-D CROSS SECTIONS          1          ***
***
***          NBK          NEUTRON BANK SIZE          1025          ***
***
***          XNB          EXTRA POSITIONS IN NEUTRON BANK          0          ***
***
***          NFB          FISSION BANK SIZE          1000          ***
***
***          XFB          EXTRA POSITIONS IN FISSION BANK          0          ***
***
***          WTA          DEFAULT VALUE OF WEIGHT AVERAGE          0.5000          ***
***
***          WTH          WEIGHT HIGH FOR SPLITTING          3.0000          ***
***
***          WTL          WEIGHT LOW FOR RUSSIAN ROULETTE          0.3333          ***
***
***          RND          STARTING RANDOM NUMBER          321          ***
***
***          NB8          NUMBER OF D.A. BLOCKS ON UNIT 8          200          ***
***
***          NL8          LENGTH OF D.A. BLOCKS ON UNIT 8          512          ***
***
***          ADJ          MODE OF CALCULATION          FORWARD          ***
***
***          INPUT DATA WRITTEN ON RESTART UNIT          NO          ***
***
***          BINARY DATA INTERFACE          YES          ***
***
*****

```

```
*****  
***  
***          TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)  
***  
*****  
***          LOGICAL PARAMETERS          ***  
***  
*** RUN   EXECUTE PROBLEM AFTER CHECKING DATA   YES           PLT   PLOT PICTURE MAP(S)             YES ***  
***  
*** FLX   COMPUTE FLUX                           NO            PDN   COMPUTE FISSION DENSITIES         NO ***  
***  
*** SMU   COMPUTE AVG UNIT SELF-MULTIPLICATION    NO            NUB   COMPUTE NU-BAR & AVG FISSION GROUP      YES ***  
***  
*** MKU   COMPUTE MATRIX K-EFF BY UNIT NUMBER     NO            MKP   COMPUTE MATRIX K-EFF BY UNIT LOCATION   NO ***  
***  
*** CKU   COMPUTE COFACTOR K-EFF BY UNIT NUMBER   NO            CKP   COMPUTE COFACTOR K-EFF BY UNIT LOCATION   NO ***  
***  
*** FMU   PRINT FISS PROD MATRIX BY UNIT NUMBER   NO            FMP   PRINT FISS PROD MATRIX BY UNIT LOCATION NO ***  
***  
*** MKH   COMPUTE MATRIX K-EFF BY HOLE NUMBER     NO            MKA   COMPUTE MATRIX K-EFF BY ARRAY NUMBER     NO ***  
***  
*** CKH   COMPUTE COFACTOR K-EFF BY HOLE NUMBER   NO            CKA   COMPUTE COFACTOR K-EFF BY ARRAY NUMBER     NO ***  
***  
*** FMH   PRINT FISS PROD MATRIX BY HOLE NUMBER   NO            FMA   PRINT FISS PROD MATRIX BY ARRAY NUMBER   NO ***  
***  
*** HHL   COLLECT MATRIX BY HIGHEST HOLE LEVEL    NO            HAL   COLLECT MATRIX BY HIGHEST ARRAY LEVEL   NO ***  
***  
*** AMX   PRINT ALL MIXED CROSS SECTIONS          NO            FAR   PRINT FIS. AND ABS. BY REGION              NO ***  
***  
*** XS1   PRINT 1-D MIXTURE X-SECTIONS            NO            GAS   PRINT FAR BY GROUP                          NO ***  
***  
*** XS2   PRINT 2-D MIXTURE X-SECTIONS            NO            PAX   PRINT XSEC-ALBEDO CORRELATION TABLES     NO ***  
***  
*** XAP   PRINT MIXTURE ANGLES & PROBABILITIES    NO            PWT   PRINT WEIGHT AVERAGE ARRAY              NO ***  
***  
*** PKI   PRINT FISSION SPECTRUM                  NO            PGM   PRINT INPUT GEOMETRY                      NO ***  
***  
*** P1D   PRINT EXTRA 1-D CROSS SECTIONS          NO            BUG   PRINT DEBUG INFORMATION                    NO ***  
***  
***                                     TRK   PRINT TRACKING INFORMATION                NO ***  
***  
*****  
  
*****  
PARAMETER INPUT COMPLETED  
  
.....      0 IO'S WERE USED READING THE PARAMETER DATA      .....
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

GENERATION	GENERATION K-EFFECTIVE	ELAPSED TIME MINUTES	AVERAGE K-EFFECTIVE	AVG K-EFF DEVIATION	MATRIX K-EFFECTIVE	MATRIX K-EFF DEVIATION
KENO MESSAGE NUMBER K5-132	WARNING....ONLY	966 INDEPENDENT	FISSION POINTS WERE	GENERATED		
1	8.86430E-01	1.63583E+00	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
2	9.30216E-01	1.64600E+00	1.00000E+00	0.00000E+00	0.00000E+00	0.00000E+00
KENO MESSAGE NUMBER K5-132	WARNING....ONLY	968 INDEPENDENT	FISSION POINTS WERE	GENERATED		
3	8.62103E-01	1.65600E+00	8.62103E-01	0.00000E+00	0.00000E+00	0.00000E+00
4	9.59846E-01	1.66517E+00	9.10975E-01	4.88714E-02	0.00000E+00	0.00000E+00
5	9.26128E-01	1.67533E+00	9.16026E-01	2.86645E-02	0.00000E+00	0.00000E+00
6	9.06831E-01	1.68533E+00	9.13727E-01	2.03988E-02	0.00000E+00	0.00000E+00
7	9.34797E-01	1.69450E+00	9.17941E-01	1.63531E-02	0.00000E+00	0.00000E+00
8	8.84345E-01	1.70450E+00	9.12342E-01	1.44788E-02	0.00000E+00	0.00000E+00
9	8.84580E-01	1.71467E+00	9.08376E-01	1.28635E-02	0.00000E+00	0.00000E+00
10	8.83476E-01	1.72383E+00	9.05263E-01	1.15667E-02	0.00000E+00	0.00000E+00
11	9.15707E-01	1.73383E+00	9.06424E-01	1.02667E-02	0.00000E+00	0.00000E+00
12	9.44775E-01	1.74383E+00	9.10259E-01	9.95150E-03	0.00000E+00	0.00000E+00
13	9.44424E-01	1.75300E+00	9.13365E-01	9.52225E-03	0.00000E+00	0.00000E+00
14	9.38542E-01	1.76400E+00	9.15463E-01	8.94221E-03	0.00000E+00	0.00000E+00
15	9.05427E-01	1.77317E+00	9.14691E-01	8.26178E-03	0.00000E+00	0.00000E+00
16	9.05457E-01	1.78333E+00	9.14031E-01	7.67731E-03	0.00000E+00	0.00000E+00
17	9.07366E-01	1.79250E+00	9.13587E-01	7.16098E-03	0.00000E+00	0.00000E+00
18	9.47593E-01	1.80150E+00	9.15712E-01	7.02758E-03	0.00000E+00	0.00000E+00
19	8.96175E-01	1.81167E+00	9.14563E-01	6.70055E-03	0.00000E+00	0.00000E+00
20	8.92676E-01	1.82167E+00	9.13347E-01	6.43330E-03	0.00000E+00	0.00000E+00
21	8.97286E-01	1.83083E+00	9.12502E-01	6.14372E-03	0.00000E+00	0.00000E+00
22	9.18874E-01	1.84100E+00	9.12820E-01	5.83715E-03	0.00000E+00	0.00000E+00
23	9.16554E-01	1.85100E+00	9.12998E-01	5.55508E-03	0.00000E+00	0.00000E+00
24	9.24305E-01	1.86117E+00	9.13512E-01	5.32144E-03	0.00000E+00	0.00000E+00
25	9.50884E-01	1.87117E+00	9.15137E-01	5.33812E-03	0.00000E+00	0.00000E+00
26	9.03892E-01	1.88033E+00	9.14668E-01	5.13230E-03	0.00000E+00	0.00000E+00
27	8.67111E-01	1.89033E+00	9.12766E-01	5.27749E-03	0.00000E+00	0.00000E+00
28	9.20762E-01	1.90050E+00	9.13074E-01	5.07977E-03	0.00000E+00	0.00000E+00
29	9.28867E-01	1.91050E+00	9.13659E-01	4.92288E-03	0.00000E+00	0.00000E+00
30	8.98241E-01	1.92067E+00	9.13108E-01	4.77566E-03	0.00000E+00	0.00000E+00
31	9.41922E-01	1.92967E+00	9.14102E-01	4.71394E-03	0.00000E+00	0.00000E+00
32	9.25394E-01	1.93983E+00	9.14478E-01	4.56963E-03	0.00000E+00	0.00000E+00
33	8.83837E-01	1.94983E+00	9.13490E-01	4.52894E-03	0.00000E+00	0.00000E+00
34	8.99376E-01	1.96000E+00	9.13049E-01	4.40725E-03	0.00000E+00	0.00000E+00
35	9.19662E-01	1.97000E+00	9.13249E-01	4.27631E-03	0.00000E+00	0.00000E+00
36	9.35925E-01	1.98017E+00	9.13916E-01	4.20189E-03	0.00000E+00	0.00000E+00
37	9.17451E-01	1.99017E+00	9.14017E-01	4.08132E-03	0.00000E+00	0.00000E+00
38	9.42446E-01	2.00017E+00	9.14807E-01	4.04418E-03	0.00000E+00	0.00000E+00
39	9.39725E-01	2.01033E+00	9.15480E-01	3.99060E-03	0.00000E+00	0.00000E+00
40	9.07176E-01	2.01950E+00	9.15262E-01	3.89031E-03	0.00000E+00	0.00000E+00
41	9.36038E-01	2.02950E+00	9.15794E-01	3.82651E-03	0.00000E+00	0.00000E+00
42	9.08300E-01	2.03967E+00	9.15607E-01	3.73432E-03	0.00000E+00	0.00000E+00
43	9.23285E-01	2.04967E+00	9.15794E-01	3.64691E-03	0.00000E+00	0.00000E+00
44	9.10821E-01	2.05967E+00	9.15676E-01	3.56099E-03	0.00000E+00	0.00000E+00
45	8.87317E-01	2.06983E+00	9.15016E-01	3.53918E-03	0.00000E+00	0.00000E+00
46	9.11245E-01	2.07983E+00	9.14931E-01	3.45887E-03	0.00000E+00	0.00000E+00
47	9.33556E-01	2.08900E+00	9.15344E-01	3.40637E-03	0.00000E+00	0.00000E+00
48	9.12917E-01	2.09917E+00	9.15292E-01	3.33192E-03	0.00000E+00	0.00000E+00
49	9.29311E-01	2.10817E+00	9.15590E-01	3.27387E-03	0.00000E+00	0.00000E+00
50	9.32855E-01	2.11833E+00	9.15950E-01	3.22506E-03	0.00000E+00	0.00000E+00
51	9.01676E-01	2.12833E+00	9.15658E-01	3.17196E-03	0.00000E+00	0.00000E+00
52	9.51940E-01	2.13850E+00	9.16384E-01	3.19147E-03	0.00000E+00	0.00000E+00
53	8.92347E-01	2.14850E+00	9.15913E-01	3.16357E-03	0.00000E+00	0.00000E+00
54	9.59600E-01	2.15867E+00	9.16753E-01	3.21389E-03	0.00000E+00	0.00000E+00
55	8.95899E-01	2.16867E+00	9.16359E-01	3.17712E-03	0.00000E+00	0.00000E+00
56	9.18557E-01	2.17783E+00	9.16400E-01	3.11800E-03	0.00000E+00	0.00000E+00
57	9.43905E-01	2.18783E+00	9.16900E-01	3.10137E-03	0.00000E+00	0.00000E+00
58	9.20120E-01	2.19800E+00	9.16958E-01	3.04603E-03	0.00000E+00	0.00000E+00
59	9.37297E-01	2.20800E+00	9.17314E-01	3.01331E-03	0.00000E+00	0.00000E+00
60	9.87268E-01	2.21817E+00	9.18521E-01	3.19713E-03	0.00000E+00	0.00000E+00
61	9.23956E-01	2.22817E+00	9.18613E-01	3.14382E-03	0.00000E+00	0.00000E+00
982	9.04672E-01	1.14153E+01	9.24700E-01	7.36990E-04	0.00000E+00	0.00000E+00
983	9.36419E-01	1.14255E+01	9.24712E-01	7.36335E-04	0.00000E+00	0.00000E+00
984	9.14772E-01	1.14355E+01	9.24702E-01	7.35654E-04	0.00000E+00	0.00000E+00

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

985	9.17418E-01	1.14447E+01	9.24694E-01	7.34943E-04	0.00000E+00	0.00000E+00
986	8.89156E-01	1.14547E+01	9.24658E-01	7.35083E-04	0.00000E+00	0.00000E+00
987	9.54606E-01	1.14648E+01	9.24689E-01	7.34966E-04	0.00000E+00	0.00000E+00
988	9.07762E-01	1.14740E+01	9.24671E-01	7.34421E-04	0.00000E+00	0.00000E+00
989	9.32998E-01	1.14840E+01	9.24680E-01	7.33725E-04	0.00000E+00	0.00000E+00
990	9.24344E-01	1.14942E+01	9.24680E-01	7.32982E-04	0.00000E+00	0.00000E+00
991	9.27070E-01	1.15042E+01	9.24682E-01	7.32244E-04	0.00000E+00	0.00000E+00
992	9.49672E-01	1.15133E+01	9.24707E-01	7.31940E-04	0.00000E+00	0.00000E+00
993	9.60024E-01	1.15233E+01	9.24743E-01	7.32069E-04	0.00000E+00	0.00000E+00
994	9.75624E-01	1.15335E+01	9.24794E-01	7.33127E-04	0.00000E+00	0.00000E+00
995	9.04812E-01	1.15435E+01	9.24774E-01	7.32664E-04	0.00000E+00	0.00000E+00
996	9.18264E-01	1.15537E+01	9.24767E-01	7.31956E-04	0.00000E+00	0.00000E+00
997	9.32982E-01	1.15628E+01	9.24776E-01	7.31267E-04	0.00000E+00	0.00000E+00
998	9.05779E-01	1.15728E+01	9.24757E-01	7.30781E-04	0.00000E+00	0.00000E+00
999	9.34645E-01	1.15828E+01	9.24767E-01	7.30115E-04	0.00000E+00	0.00000E+00
1000	9.34634E-01	1.15930E+01	9.24776E-01	7.29450E-04	0.00000E+00	0.00000E+00
1001	8.95950E-01	1.16022E+01	9.24748E-01	7.29291E-04	0.00000E+00	0.00000E+00
1002	9.32221E-01	1.16122E+01	9.24755E-01	7.28600E-04	0.00000E+00	0.00000E+00
1003	9.03383E-01	1.16223E+01	9.24734E-01	7.28184E-04	0.00000E+00	0.00000E+00

KENO MESSAGE NUMBER K5-123

EXECUTION TERMINATED DUE TO COMPLETION OF THE SPECIFIED NUMBER OF GENERATIONS.

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

LIFETIME = 4.29740E-05 + OR - 8.70199E-08 GENERATION TIME = 2.88492E-05 + OR - 4.67432E-08
 NU BAR = 2.44142E+00 + OR - 6.38806E-05 AVERAGE FISSION GROUP = 2.17829E+01 + OR - 3.80771E-03
 ENERGY (EV) OF THE AVERAGE LETHARGY CAUSING FISSION = 2.67732E-01 + OR - 8.25407E-04

NO. OF INITIAL GENERATIONS SKIPPED	AVERAGE K-EFFECTIVE	DEVIATION	67 PER CENT CONFIDENCE INTERVAL	95 PER CENT CONFIDENCE INTERVAL	99 PER CENT CONFIDENCE INTERVAL	NUMBER OF HISTORIES
3	0.92480	+ OR - 0.00073	0.92407 TO 0.92552	0.92334 TO 0.92625	0.92262 TO 0.92698	1000000
4	0.92476	+ OR - 0.00073	0.92404 TO 0.92549	0.92331 TO 0.92621	0.92258 TO 0.92694	999000
5	0.92476	+ OR - 0.00073	0.92403 TO 0.92549	0.92331 TO 0.92621	0.92258 TO 0.92694	998000
6	0.92478	+ OR - 0.00073	0.92405 TO 0.92551	0.92332 TO 0.92623	0.92260 TO 0.92696	997000
7	0.92477	+ OR - 0.00073	0.92404 TO 0.92550	0.92331 TO 0.92622	0.92258 TO 0.92695	996000
8	0.92481	+ OR - 0.00073	0.92408 TO 0.92554	0.92335 TO 0.92626	0.92263 TO 0.92699	995000
9	0.92485	+ OR - 0.00073	0.92412 TO 0.92558	0.92339 TO 0.92630	0.92267 TO 0.92703	994000
10	0.92489	+ OR - 0.00073	0.92416 TO 0.92562	0.92344 TO 0.92634	0.92271 TO 0.92707	993000
11	0.92490	+ OR - 0.00073	0.92417 TO 0.92563	0.92345 TO 0.92635	0.92272 TO 0.92708	992000
12	0.92488	+ OR - 0.00073	0.92415 TO 0.92561	0.92342 TO 0.92634	0.92270 TO 0.92706	991000
17	0.92490	+ OR - 0.00073	0.92417 TO 0.92563	0.92344 TO 0.92636	0.92271 TO 0.92709	986000
22	0.92498	+ OR - 0.00073	0.92424 TO 0.92571	0.92351 TO 0.92644	0.92278 TO 0.92717	981000
27	0.92504	+ OR - 0.00073	0.92431 TO 0.92577	0.92358 TO 0.92650	0.92284 TO 0.92724	976000
32	0.92505	+ OR - 0.00074	0.92432 TO 0.92579	0.92358 TO 0.92652	0.92284 TO 0.92726	971000
37	0.92512	+ OR - 0.00074	0.92438 TO 0.92586	0.92365 TO 0.92660	0.92291 TO 0.92733	966000
42	0.92511	+ OR - 0.00074	0.92437 TO 0.92585	0.92363 TO 0.92659	0.92289 TO 0.92733	961000
47	0.92518	+ OR - 0.00074	0.92443 TO 0.92592	0.92369 TO 0.92666	0.92295 TO 0.92740	956000
912	0.92641	+ OR - 0.00227	0.92414 TO 0.92868	0.92186 TO 0.93096	0.91959 TO 0.93323	91000
917	0.92684	+ OR - 0.00235	0.92449 TO 0.92919	0.92214 TO 0.93154	0.91979 TO 0.93388	86000
922	0.92810	+ OR - 0.00235	0.92574 TO 0.93045	0.92339 TO 0.93281	0.92104 TO 0.93516	81000
927	0.92697	+ OR - 0.00239	0.92458 TO 0.92937	0.92218 TO 0.93176	0.91979 TO 0.93416	76000
932	0.92664	+ OR - 0.00250	0.92414 TO 0.92915	0.92163 TO 0.93165	0.91913 TO 0.93416	71000
937	0.92719	+ OR - 0.00264	0.92455 TO 0.92983	0.92190 TO 0.93248	0.91926 TO 0.93512	66000
942	0.92638	+ OR - 0.00282	0.92356 TO 0.92920	0.92074 TO 0.93202	0.91792 TO 0.93484	61000
947	0.92701	+ OR - 0.00302	0.92399 TO 0.93003	0.92098 TO 0.93305	0.91796 TO 0.93607	56000
952	0.92705	+ OR - 0.00320	0.92385 TO 0.93025	0.92065 TO 0.93345	0.91745 TO 0.93665	51000
957	0.92734	+ OR - 0.00349	0.92386 TO 0.93083	0.92037 TO 0.93432	0.91688 TO 0.93781	46000
962	0.92770	+ OR - 0.00357	0.92413 TO 0.93127	0.92055 TO 0.93485	0.91698 TO 0.93842	41000
967	0.92517	+ OR - 0.00374	0.92143 TO 0.92891	0.91770 TO 0.93264	0.91396 TO 0.93638	36000
972	0.92346	+ OR - 0.00404	0.91942 TO 0.92749	0.91538 TO 0.93153	0.91135 TO 0.93557	31000
977	0.92528	+ OR - 0.00416	0.92112 TO 0.92945	0.91696 TO 0.93361	0.91279 TO 0.93778	26000

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

982	0.92631	+ OR - 0.00479	0.92153 TO 0.93110	0.91674 TO 0.93588	0.91196 TO 0.94067	21000
987	0.92751	+ OR - 0.00544	0.92207 TO 0.93295	0.91663 TO 0.93839	0.91120 TO 0.94383	16000
992	0.92712	+ OR - 0.00749	0.91963 TO 0.93461	0.91213 TO 0.94210	0.90464 TO 0.94960	11000
997	0.91777	+ OR - 0.00731	0.91045 TO 0.92508	0.90314 TO 0.93240	0.89583 TO 0.93971	6000

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

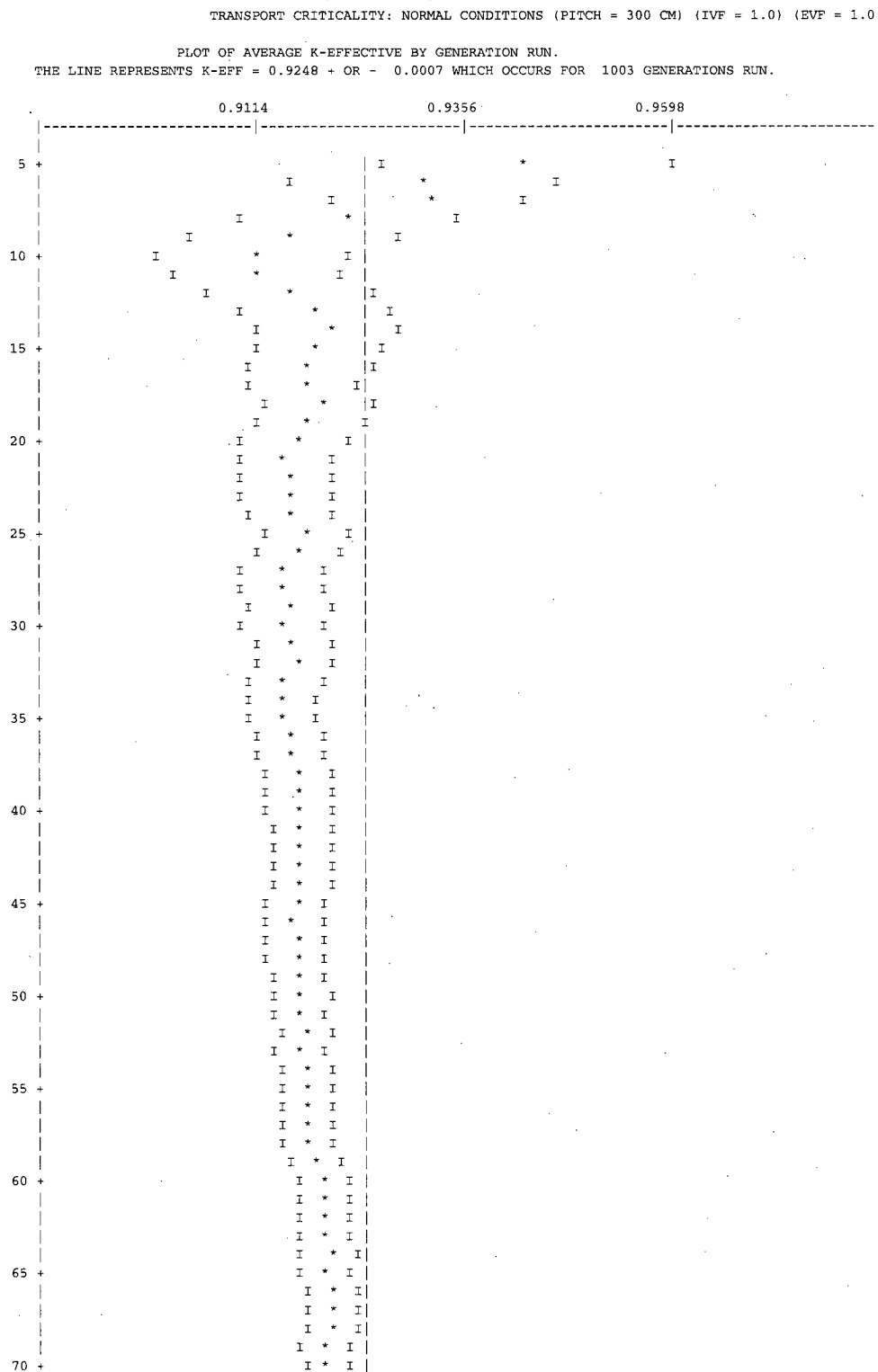


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

	I * I
	I * I
	I * I
	I * I
75 +	I * I
	I * I
	I * I
	I * I
80 +	I * I
	I * I
	I * I
	I * I
	I * I
85 +	I * I
	I * I
	I * I
	I * I
90 +	I * I
	I * I
	I * I
	I * I
	I * I
95 +	I * I
	I * I
	I * I
	I * I
	I * I
100 +	I * I
	I * I
	I * I
	I * I
	I * I
105 +	I * I
	I * I
	I * I
	I * I
	I * I
110 +	I * I
	I * I
	I * I
	I * I
	I * I
115 +	I * I
	I * I
	I * I
	I * I
	I * I
120 +	I * I
	I * I
	I * I
	I * I
	I * I
125 +	I * I
	I * I
	I * I
	I * I
	I * I
130 +	I * I
	I * I
	I * I
	I * I
	I * I
135 +	I * I
	I * I
	I * I
	I * I
	I * I
140 +	I * I
	I * I
	I * I
	I * I
	I * I

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
145 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
150 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
155 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
160 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
165 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
170 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
175 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
180 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
185 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
190 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
195 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
200 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
205 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
210 + I * I
      I * I
      I * I
      I * I
      I * I
      I * I
215 + I * I
      I * I
      I * I
      I * I
```

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

```
220 + I|*I
      I|*I
      I|*I
      I* I
      I|*I
      I|*I
225 + I|*I
      I|*I
      I|*I
      I* I
      I* I
      I* I
      I* I
230 + I* I
      I* I
      I* I
      I* I
      I* I
235 + I* I
      I* I
      I|*I
      I|*I
      I|*I
240 + I|*I
      I|*I
      I|*I
      I|*I
      I|*I
245 + I|*I
      I|*I
      I|*I
      I|*I
      I* I
250 + I|*I
      I|*I
      I|*I
      I* I
      I* I
255 + I *I
      I *I
      I *I
      I *I
      I *I
260 + I *I
      I *I
      I *I
      I *I
      I *I
265 + I *I
      I *I
      I *I
      I *I
      I *I
270 + }I* I
      }I* I
      }I* I
      }I* I
      }I* I
275 + }I* I
      }I* I
      I *I

905 + I*I
      I*I
      I*I
      I*I
      I*I
910 + I*I
      I*I
      I*I
      I*I
      I*I
915 + I*I
      I*I
      I*I
```

	I*I
	I*I
920 +	I*I
	I*I
	I*I
	I*I
	I*I
	I*I
925 +	I*I
	I*I
	I*I
	I*I
	I*I
930 +	I*I
	I*I
	I*I
	I*I
	I*I
935 +	I*I
	I*I
	I*I
	I*I
	I*I
940 +	I*I
	I*I
	I*I
	I*I
	I*I
945 +	I*I
	I*I
	I*I
	I*I
	I*I
950 +	I*I
	I*I
	I*I
	I*I
	I*I
955 +	I*I
	I*I
	I*I
	I*I
	I*I
960 +	I*I
	I*I
	I*I
	I*I
	I*I
965 +	I*I
	*I
	*I
	*I
	*I
970 +	*I
	*I
	*I
	*I
	*I
975 +	I*I
	*I
	*I
	*I
	*I
980 +	*I
	*I
	*I
	*I
985 +	*I
	I*I
	*I
	I*I
	I*I
990 +	I*I
	I*I

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

	*I
	*I
	*I
995 +	*I
	*I
	*I
	*I
	*I
1000 +	*I
	*I
	*I
	*I

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

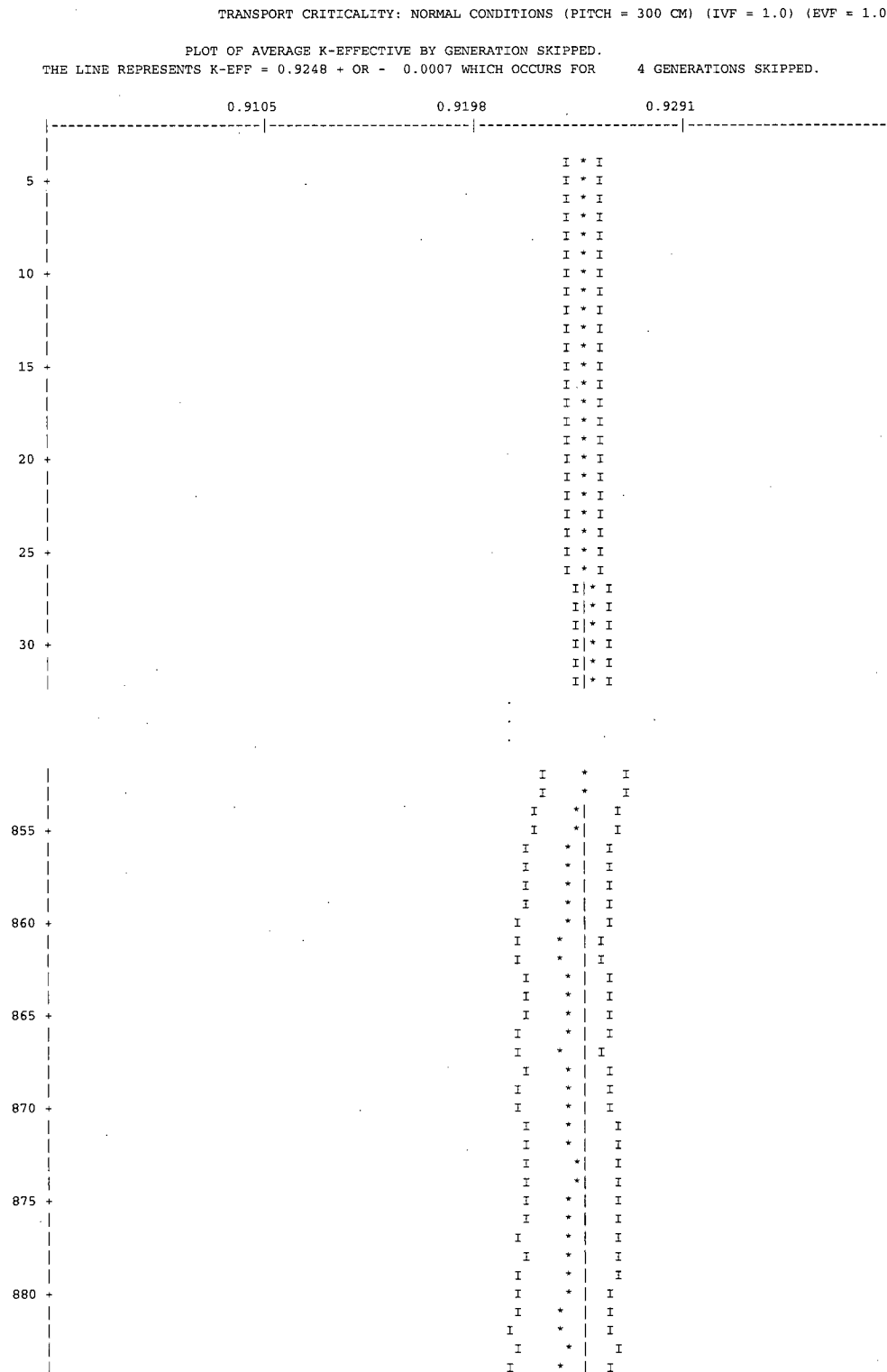


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

885 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
890 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
895 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
900 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
905 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
910 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
915 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
920 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
925 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
930 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
935 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
940 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
945 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
950 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I
955 +	I	*	I
	I	*	I
	I	*	I
	I	*	I
	I	*	I

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

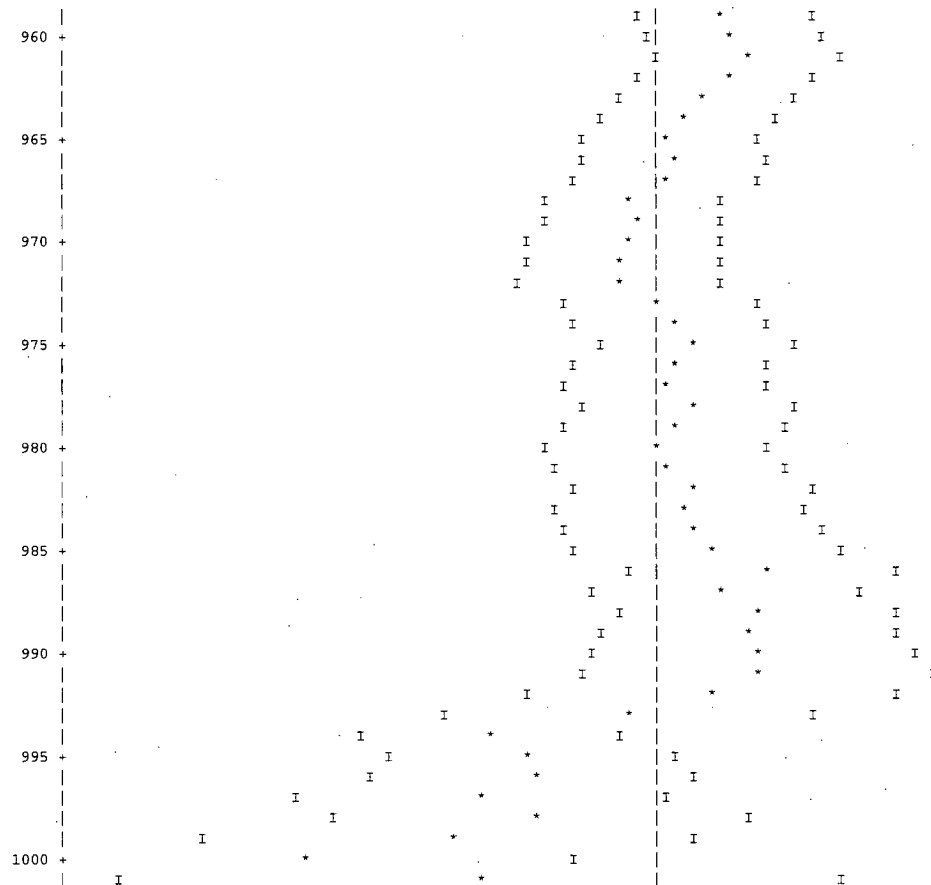


Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

	FREQUENCY FOR GENERATIONS	4 TO 1003
0.8469 TO 0.8499	*	
0.8499 TO 0.8529		
0.8529 TO 0.8560	*	
0.8560 TO 0.8590		
0.8590 TO 0.8620	*	
0.8620 TO 0.8650		
0.8650 TO 0.8680	**	
0.8680 TO 0.8710	****	
0.8710 TO 0.8741	****	
0.8741 TO 0.8771	*	
0.8771 TO 0.8801	*****	
0.8801 TO 0.8831	*****	
0.8831 TO 0.8861	*****	
0.8861 TO 0.8891	*****	
0.8891 TO 0.8922	*****	
0.8922 TO 0.8952	*****	
0.8952 TO 0.8982	*****	
0.8982 TO 0.9012	*****	
0.9012 TO 0.9042	*****	
0.9042 TO 0.9072	*****	
0.9072 TO 0.9103	*****	
0.9103 TO 0.9133	*****	
0.9133 TO 0.9163	*****	
0.9163 TO 0.9193	*****	
0.9193 TO 0.9223	*****	
0.9223 TO 0.9253	*****	
0.9253 TO 0.9284	*****	
0.9284 TO 0.9314	*****	
0.9314 TO 0.9344	*****	
0.9344 TO 0.9374	*****	
0.9374 TO 0.9404	*****	
0.9404 TO 0.9434	*****	
0.9434 TO 0.9465	*****	
0.9465 TO 0.9495	*****	
0.9495 TO 0.9525	*****	
0.9525 TO 0.9555	*****	
0.9555 TO 0.9585	*****	
0.9585 TO 0.9615	*****	
0.9615 TO 0.9646	*****	
0.9646 TO 0.9676	*****	
0.9676 TO 0.9706	****	
0.9706 TO 0.9736	*****	
0.9736 TO 0.9766	*****	
0.9766 TO 0.9796		
0.9796 TO 0.9827	****	
0.9827 TO 0.9857	**	
0.9857 TO 0.9887	*	
0.9887 TO 0.9917	**	
0.9917 TO 0.9947		
0.9947 TO 0.9977		
0.9977 TO 1.0008	*	

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 254 TO 1003

0.8469 TO 0.8499	*
0.8499 TO 0.8529	
0.8529 TO 0.8560	*
0.8560 TO 0.8590	
0.8590 TO 0.8620	*
0.8620 TO 0.8650	
0.8650 TO 0.8680	*
0.8680 TO 0.8710	****
0.8710 TO 0.8741	****
0.8741 TO 0.8771	*
0.8771 TO 0.8801	*****
0.8801 TO 0.8831	*****
0.8831 TO 0.8861	*****
0.8861 TO 0.8891	*****
0.8891 TO 0.8922	*****
0.8922 TO 0.8952	*****
0.8952 TO 0.8982	*****
0.8982 TO 0.9012	*****
0.9012 TO 0.9042	*****
0.9042 TO 0.9072	*****
0.9072 TO 0.9103	*****
0.9103 TO 0.9133	*****
0.9133 TO 0.9163	*****
0.9163 TO 0.9193	*****
0.9193 TO 0.9223	*****
0.9223 TO 0.9253	*****
0.9253 TO 0.9284	*****
0.9284 TO 0.9314	*****
0.9314 TO 0.9344	*****
0.9344 TO 0.9374	*****
0.9374 TO 0.9404	*****
0.9404 TO 0.9434	*****
0.9434 TO 0.9465	*****
0.9465 TO 0.9495	*****
0.9495 TO 0.9525	*****
0.9525 TO 0.9555	*****
0.9555 TO 0.9585	*****
0.9585 TO 0.9615	*****
0.9615 TO 0.9646	*****
0.9646 TO 0.9676	*****
0.9676 TO 0.9706	****
0.9706 TO 0.9736	****
0.9736 TO 0.9766	*****
0.9766 TO 0.9796	
0.9796 TO 0.9827	***
0.9827 TO 0.9857	**
0.9857 TO 0.9887	
0.9887 TO 0.9917	
0.9917 TO 0.9947	
0.9947 TO 0.9977	
0.9977 TO 1.0008	*

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 504 TO 1003

0.8469 TO 0.8499	*
0.8499 TO 0.8529	
0.8529 TO 0.8560	*
0.8560 TO 0.8590	
0.8590 TO 0.8620	
0.8620 TO 0.8650	
0.8650 TO 0.8680	*
0.8680 TO 0.8710	****
0.8710 TO 0.8741	***
0.8741 TO 0.8771	*
0.8771 TO 0.8801	****
0.8801 TO 0.8831	**
0.8831 TO 0.8861	*****
0.8861 TO 0.8891	*****
0.8891 TO 0.8922	*****
0.8922 TO 0.8952	*****
0.8952 TO 0.8982	*****
0.8982 TO 0.9012	*****
0.9012 TO 0.9042	*****
0.9042 TO 0.9072	*****
0.9072 TO 0.9103	*****
0.9103 TO 0.9133	*****
0.9133 TO 0.9163	*****
0.9163 TO 0.9193	*****
0.9193 TO 0.9223	*****
0.9223 TO 0.9253	*****
0.9253 TO 0.9284	*****
0.9284 TO 0.9314	*****
0.9314 TO 0.9344	*****
0.9344 TO 0.9374	*****
0.9374 TO 0.9404	*****
0.9404 TO 0.9434	*****
0.9434 TO 0.9465	*****
0.9465 TO 0.9495	*****
0.9495 TO 0.9525	*****
0.9525 TO 0.9555	*****
0.9555 TO 0.9585	*****
0.9585 TO 0.9615	*****
0.9615 TO 0.9646	*****
0.9646 TO 0.9676	*****
0.9676 TO 0.9706	****
0.9706 TO 0.9736	***
0.9736 TO 0.9766	****
0.9766 TO 0.9796	
0.9796 TO 0.9827	*
0.9827 TO 0.9857	*
0.9857 TO 0.9887	
0.9887 TO 0.9917	
0.9917 TO 0.9947	
0.9947 TO 0.9977	
0.9977 TO 1.0008	*

Figure 6.7-11 CSAS Input/Output Summary for Damaged Fuel Can Configuration – Normal Conditions (continued)

TRANSPORT CRITICALITY: NORMAL CONDITIONS (PITCH = 300 CM) (IVF = 1.0) (EVF = 1.0)

FREQUENCY FOR GENERATIONS 754 TO 1003

0.8469 TO 0.8499	*
0.8499 TO 0.8529	
0.8529 TO 0.8560	
0.8560 TO 0.8590	
0.8590 TO 0.8620	
0.8620 TO 0.8650	
0.8650 TO 0.8680	
0.8680 TO 0.8710	***
0.8710 TO 0.8741	*
0.8741 TO 0.8771	
0.8771 TO 0.8801	****
0.8801 TO 0.8831	**
0.8831 TO 0.8861	****
0.8861 TO 0.8891	**
0.8891 TO 0.8922	*****
0.8922 TO 0.8952	*****
0.8952 TO 0.8982	*****
0.8982 TO 0.9012	*****
0.9012 TO 0.9042	*****
0.9042 TO 0.9072	*****
0.9072 TO 0.9103	*****
0.9103 TO 0.9133	*****
0.9133 TO 0.9163	*****
0.9163 TO 0.9193	*****
0.9193 TO 0.9223	*****
0.9223 TO 0.9253	*****
0.9253 TO 0.9284	*****
0.9284 TO 0.9314	*****
0.9314 TO 0.9344	*****
0.9344 TO 0.9374	*****
0.9374 TO 0.9404	*****
0.9404 TO 0.9434	*****
0.9434 TO 0.9465	*****
0.9465 TO 0.9495	*****
0.9495 TO 0.9525	*****
0.9525 TO 0.9555	*****
0.9555 TO 0.9585	*****
0.9585 TO 0.9615	***
0.9615 TO 0.9646	**
0.9646 TO 0.9676	***
0.9676 TO 0.9706	*
0.9706 TO 0.9736	**
0.9736 TO 0.9766	****
0.9766 TO 0.9796	
0.9796 TO 0.9827	*
0.9827 TO 0.9857	
0.9857 TO 0.9887	
0.9887 TO 0.9917	
0.9917 TO 0.9947	
0.9947 TO 0.9977	
0.9977 TO 1.0008	

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7.0 OPERATING PROCEDURES

This chapter provides an outline of the operating procedures and tests that are performed to ensure proper function of the NAC-STC during transport operations. The operating procedures provided in this chapter are the minimum generic requirements for loading, unloading, preparation for transport, and for inspection and testing of the cask. Bolt torque values are provided in Table 7-1. Each licensee and cask user will develop, prepare and approve site specific procedures, based on the approved detailed operating procedures provided by NAC, to assure that cask handling and shipping activities are performed in accordance with the package Certificate of Compliance and the applicable Nuclear Regulatory Commission and Department of Transportation regulations governing the packaging and transport of radioactive materials.

These procedures assume that the unloaded NAC-STC arrives at a site already configured for use at the site. If this is not the case, then additional operations would be specified in the site specific procedures to configure the cask for the intended use.

The operating procedures in this chapter have been written assuming direct loading or unloading of fuel in the basket in the NAC-STC in a spent fuel pool, or dry loading and unloading of a sealed canister in the reactor cask receiving area, fuel building or other suitable location identified by the user. With minor modifications, site-specific procedures can be written to accommodate the dry direct loading or unloading of fuel from the cask in a hot cell.

Procedures are also provided for the preparation for shipment of an NAC-STC cask that has been loaded and stored at an Independent Spent Fuel Storage Installation (ISFSI) in accordance with the ISFSI license and the 10 CFR 72 requirements.

It is the responsibility of the cask user to prepare site-specific handling procedures in accordance with the Certificate of Compliance, these generic procedures, and the licensee's Quality Assurance program. User approved operating procedures ensure that critical steps are not overlooked, that the packaging is handled in accordance with its Certificate of Compliance and Safety Analysis Report.

The user will verify by fuel accounting, historical data, and inspection records, that the fuel assemblies to be loaded are in compliance with the content conditions of the Certificate of Compliance. In the directly loaded configuration, fuel assemblies or fuel rods with known or suspected cladding defects that exceed pin holes and hairline cracks are not to be loaded into the

NAC-STC. In the canistered configuration, damaged (failed) fuel will be separately containerized (canned) and sealed in the canister prior to transport.

The user shall verify that the NAC-STC transport cask has the correct O-ring configuration for the intended use. The transport cask may be configured with either metallic O-rings or with non-metallic Viton O-rings. The O-rings may not be used interchangeably, since each O-ring type requires a different lid O-ring groove configuration. Consequently, the inner lid, vent and drain port coverplates and outer lid are machined with a square O-ring groove to accept metallic O-rings or are machined with a truncated triangular (dove-tail) groove to accept Viton O-rings.

Viton O-rings may be used only when directly loading spent fuel for transport without interim storage. Metallic O-rings must be used when directly loading spent fuel for an extended period of storage and may be used when directly loading spent fuel for transport without interim storage. Metallic O-rings must also be used when loading canistered fuel or GTCC waste for transport. The metallic and nonmetallic O-rings have different limits of allowable leak rate as specified in the procedures.

Table 7-1 Torque Table

Component	No. Used	Fastener ¹	Torque Value ²
Outer Lid Bolt	36	1-8 - UNC Socket Head Cap Screw	550 ± 50 ft.-lb (746 ± 68 N-m)
Inner Lid Bolt	42	1 ½ - 8 UN-2A Socket Head Cap Screw	2,540 ± 200 ft-lb (3,443 ± 271 N-m)
Port Cover Bolt	6	¾ - 16 UNC Socket Head Cap Screw	140 ± 10 in-lb (16 ± 1 N-m)
Coverplate Bolt	8	½ - 13 UNC Socket Head Cap Screw	300 ± 20 in-lb (34 ± 2 N-m)
Test Plug	1	Part No. 423-803-13	30 ± 3 ft-lb (41 ± 4 N-m)
Test Plug	2	Part No. 423-806-3	70 ± 5 in-lb (8 ± 0.6 N-m)
Test Plug	2	Part No. 423-807-8	70 ± 5 in-lb (8 ± 0.6 N-m)
Redwood Impact Limiter Retaining Rods	32	Part No. 423-811-7	75 ± 5 ft-lb (102 ± 7 N-m)
Balsa Impact Limiter Retaining Rods	32	Part No. 423-859-1	75 ± 5 ft-lb (102 ± 7 N-m)
Impact Limiter Nut	32	1 - 8 UNC - 2B Heavy Hex Nut	35 ± 2 ft-lb (47 ± 3 N-m)
Impact Limiter Jam Nut	32	1 - 8 UN - 2B Heavy Hex Nut	75 ± 5 ft-lb (102 ± 7 N-m)
Adapter Ring	3	1 ½ -8UN-2A Socket Head Cap Screw	100 ± 20 ft-lb (136 ± 27 N-m)

- 1 Torque values for fasteners not shown are provided on the appropriate license drawing in Section 1.3.2.
- 2 All threaded fasteners shall be lightly lubricated using Nuclear Grade Pure Nickel NEVER-SEEZ® or equivalent.

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7.1 Outline of Procedures for Receipt and Loading the Cask

The following receipt and loading procedures are based on an acceptable cask receipt inspection for first time loading with spent fuel. For casks previously loaded and transported, the receiving inspections will require performance of radiation and removable contamination surveys of the empty cask and vehicle in accordance with 10 CFR 71 and 49 CFR 173 in the U.S. Similar requirements are contained in IAEA TS-R-1.

7.1.1 Receiving Inspection

1. Perform radiation and removable contamination surveys in accordance with 49 CFR 173.441 and 173.443 requirements.
2. Move the transport vehicle with the cask to the cask receiving area.
3. Secure the transport vehicle. Remove the personnel barrier hold down bolts from both sides of the personnel barrier. Using the lifting sling, lift the personnel barrier off of the cask and store it in a designated area.
4. Visually inspect the NAC-STC while secured to the transport vehicle in the horizontal orientation for any signs of damage.
5. Attach slings to the top impact limiter lifting points, remove impact limiter lock wires, impact limiter jam nuts, impact limiter nuts and retaining rods. Remove impact limiter and store upright. Repeat operation for the bottom impact limiter.
6. Release the tiedown assembly from the front support by removing the front tiedown bolts and lock washers.
7. Attach a sling to the tiedown assembly lifting eyes and remove the tiedown assembly from the transport vehicle.
8. Attach the cask lifting yoke to a crane hook with the appropriate load rating. Engage the two yoke arms with the lifting trunnions at the top (front) end of the cask. Rotate/lift the cask to the vertical orientation and raise the cask off of the blocks of the rear support structure of the transport vehicle. Place the cask in the vertical orientation in a decontamination area or other suitable location identified by the user. Disengage the cask lifting yoke from the lifting trunnions.

7.1.2 Preparation of Cask for Loading

The loading procedures are based on the assumption that the cask is being prepared for first time fuel loading following fabrication, or that the scheduled annual maintenance required by the

Certificate of Compliance has been successfully completed within the previous 12 months. If the cask has been used previously, at the start of this procedure, the cask is assumed to be externally decontaminated, empty of fuel contents, and sitting in the decontamination area, or in another location convenient for preparing the cask.

There are two (2) loading options for the NAC-STC. Each requires different preparation steps. The first is direct loading of fuel assemblies into a fuel basket installed in the cask, which is typically performed under water in the spent fuel pool cask loading area. The second is dry loading of a welded transportable storage canister that is already loaded with spent fuel assemblies, Reconfigured Fuel Assemblies, damaged fuel in damaged fuel cans, or with Greater Than Class C (GTCC) waste. Dry loading of the canister into the cask is performed in the cask receiving area, or another convenient location established by the user, using a transfer cask system. This section presents the generic procedures used to prepare the cask for loading for both wet direct fuel loading or dry canister loading.

7.1.2.1 Preparation for Direct Fuel Loading (Uncanistered)

This procedure presents the steps necessary to prepare the cask for under water direct loading of fuel into a basket contained in the NAC-STC cask. This procedure may be modified to accommodate the dry direct loading of fuel in a hot cell.

1. Install appropriate work platforms/scaffolding to allow access to the top of the cask.
2. Detorque in reverse torquing sequence and remove the outer lid bolts. Install the two outer lid alignment pins.
3. Install lifting eyes in the outer lid lifting holes and attach the outer lid lifting sling to the outer lid and overhead crane. Remove the outer lid and place it aside in a temporary storage area. When setting the outer lid down, protect the O-ring and the O-ring groove of the lid from damage. Remove the outer lid alignment pins. Decontaminate the surface of the inner lid and top forging as required. At a convenient time, if a metallic O-ring is used, remove and replace the metallic O-ring in the outer lid. If a Viton O-ring is used, inspect the O-ring and replace as necessary.
4. Detorque drain and vent coverplate bolts and remove the drain port and the vent port coverplates from the inner lid. Store in temporary storage area.

5. Connect demineralized water supply to drain port quick-disconnect. Connect vent hose to vent port quick-disconnect. Fill cask using demineralized water supply until water discharges from the vent hose. Ensure that the vent hose discharges into an appropriate rad waste handling system, as the cask interior may contain residual contamination.
6. Detorque and remove two inner lid bolts and install the two inner lid alignment pins at locations marked on the inner lid.
7. Detorque and remove the remaining inner lid bolts. Clean and visually inspect the outer lid bolts, inner lid bolts, and coverplate bolts for damage or excessive wear.
8. Detorque and remove the bolts and the interlid port and pressure port covers from the top forging. Store and protect all removed parts.
9. Attach the lifting yoke to a crane hook with the appropriate load rating and engage the yoke arms with the lifting trunnions.
10. Attach the lifting eyes to the inner lid. Install the inner lid lifting sling to the eyes in the inner lid and to the lifting eyes on the strongbacks of the lifting yoke.
11. Move the cask to the pool over the cask loading area. As the cask is lowered onto the cask loading area in the pool, spray the external surface of the cask with clean demineralized water to minimize external decontamination efforts.
12. After the cask is resting on the floor of the pool, disconnect the lifting yoke from the lifting trunnions and slowly raise the yoke to remove the inner lid.
13. Remove the lifting yoke and inner lid from the pool. Spray the yoke and lid, as they come out of the water to remove contamination.
14. Store the inner lid in a temporary storage area; remove and store the yoke and inner lid lifting sling in the storage area. When setting the inner lid down, ensure that the O-rings and O-ring grooves of the lid are protected from damage. Decontaminate inner lid, as necessary. At a convenient time, if metallic O-rings are used, remove and replace the metallic O-rings in the inner lid and in the vent and drain port coverplates. If Viton O-rings are used, inspect the O-rings and replace as necessary.
15. Visually examine the internal cavity, fuel basket and drain line to ensure that: (a) no damage has occurred during transit; (b) no foreign materials are present that would inhibit cavity draining; and (c) all required components are in place.

7.1.2.2 Preparation for Canister Loading

This procedure presents the steps required for loading canistered fuel or canistered GTCC waste into the NAC-STC. A canister of fuel or of GTCC waste is loaded dry into the cask, using a

transfer cask and attendant support hardware. Configuration control of the NAC-STC is required to ensure the cask has the correct impact limiters and internal spacer(s) for the canister that is to be transported. The operation of the transfer cask is described in NAC approved site specific procedures. Loading of canistered fuel or canistered GTCC waste into the NAC-STC is done in the cask receiving area, or other suitable location specified by the user. The NAC-STC is assumed to be positioned in the area designated for dry canister loading and configured with metallic O-rings.

1. Install appropriate work platforms/scaffolding to allow access to the top of the cask.
2. Detorque in reverse torquing sequence and remove the outer lid bolts. Install the two outer lid alignment pins.
3. Install lifting eyes in the outer lid lifting holes and attach the outer lid lifting device to the outer lid and overhead crane. Remove the outer lid and place it aside in a temporary storage area. When storing the outer lid, protect the O-ring and the O-ring groove of the lid from damage. Remove the outer lid alignment pins. Decontaminate the surface of the inner lid and top forging as required.
4. Detorque the vent and drain coverplate bolts and remove the drain port coverplate and the vent port coverplate from the inner lid. Store the coverplates and bolts in a designated temporary storage area.
5. Detorque and remove two inner lid bolts and install the two inner lid alignment pins at locations marked on the inner lid.
6. Attach the inner lid lifting eyebolts and the inner lid lifting slings to the inner lid.
7. Detorque and remove the remaining inner lid bolts. Clean and visually inspect the outer lid bolts, inner lid bolts, and coverplate bolts for damage or excessive wear. Record inspection results on cask loading report. Replace damaged bolts with approved spare parts.
8. Detorque and remove the bolts and covers from the interlid port and the pressure port in the top forging. Store and protect all removed parts.
9. Lower auxiliary hook to above inner lid and engage lid lifting sling to auxiliary crane hook.
10. Slowly lift and remove the inner lid. The inner lid alignment pins will guide the inner lid until it clears the top forging.
11. Store the inner lid in a temporary storage area. When storing the inner lid, ensure that the O-rings and O-ring grooves of the lid are protected from damage. Decontaminate the inner lid, as necessary.

12. Visually examine the internal cavity to ensure that the cavity is free of damage and foreign materials.
13. Install the appropriate bottom spacer for the canister to be loaded. Attach the spacer lift fixture to the spacer. Using a suitable crane, lower the spacer into the cask cavity, and remove the lift fixture.
14. Install the adapter ring and torque the three bolts to 100 ± 20 ft.-lb.
15. Install the transfer cask adapter plate on top of the NAC-STC cask.

7.1.3 Loading the NAC-STC Cask

There are three loading options for the NAC-STC cask, with each requiring different steps. The first is direct loading of fuel assemblies for transport without interim storage and the second option is for transport after a period of interim storage. These loading configurations are assumed to be performed under water in the spent fuel pool cask loading area. The third option is dry loading into the cask of a sealed transportable storage canister that already contains spent fuel or GTCC waste. Dry loading of the canister into the cask is performed in the cask receiving area, or other convenient location established by the user, using a transfer cask. This section presents the generic loading procedures for these options. In both cases, the fuel assemblies to be directly loaded, or those contained within the sealed canister, must conform to the NAC-STC Certificate of Compliance.

7.1.3.1 Direct Loading of Fuel (Uncanistered)

The NAC-STC may be closed with either metallic or nonmetallic O-rings in the containment boundary and outer lid. Metallic O-rings are required: 1) when directly loading spent fuel for an extended period of storage; and 2) when loading canistered fuel or GTCC waste (for transport). Metallic O-rings or Viton O-rings may be used when directly loading spent fuel for transport without interim storage. However, the metallic and non-metallic O-rings may not be used interchangeably, as the O-ring grooves in the lids and port covers are different for each O-ring type. As specified in the appropriate steps of this procedure, the two types of O-rings have different allowable leak rates so the lid and O-ring configurations to be used must be confirmed and the associated leak test requirements identified.

1. Using approved fuel identification and handling procedures and fuel handling equipment, engage the fuel handling tool to the top of the fuel assembly, lift it from the storage rack location, transfer it to above the cask, and carefully lower it into the

designated location in the fuel basket. Be careful not to contact any of the sealing surfaces on the top forging, or to come in contact with the inner lid guide pins during fuel assembly movement.

Note: Each fuel assembly shall contain the standard number of fuel rods for an assembly of that type. Dummy rods of equivalent water displacement must be substituted for removed fuel rods.

2. Record in the cask loading report the fuel identification number and basket position where the fuel assembly was placed.
3. Repeat steps 1 and 2 until the basket is fully loaded or until all desired fuel assemblies have been loaded. If the cask is going to be partially loaded, the fuel assemblies should be loaded, if possible, in a fully symmetric pattern to ensure that the center of gravity of the cask remains aligned as close as possible to the longitudinal axis of the cask.
4. Attach the inner lid lifting sling to an auxiliary crane hook and lift the inner lid. For the Viton O-ring assembly, inspect the O-ring and replace if damaged. For the metallic O-ring assembly, remove the inner lid O-rings, clean the groove surfaces, and install new metallic O-rings. Inspect new O-rings for damage prior to installation. Secure the metallic O-rings in the groove by the use of the O-ring clips and screws. Similarly, replace the metallic O-rings in the vent and drain port coverplates, or inspect the Viton O-rings and replace if required.
5. After replacing the inner lid O-rings, lift the inner lid and place it on the cask using the inner lid alignment pins to assist in proper lid seating and orientation. Visually verify proper lid position.
6. Disconnect the lid lifting device from the auxiliary crane hook and remove crane hook from area.
7. Attach the lifting yoke to the crane hook, lower the lifting yoke into the lifting position over the cask lifting trunnions, and engage the lifting arms to the lifting trunnions. Slowly lift the cask out of the pool until the top of the cask is slightly above the pool water level.

Note: As an alternative method, the cask and inner lid may be handled simultaneously.

In the event that this method is chosen, instead of performing steps 5, 6 and 7, attach the lifting yoke to a crane hook and the inner lid lifting eyes to the lift yoke. Lower the lid and engage to the cask using the lid alignment pins. Engage lifting arms to lifting trunnions. Slowly lift the cask out of the pool until the top of the cask is slightly above the pool water level.

8. Attach a drain line to the quick-disconnect in the interlid port (located in the top forging) and allow the water to drain from the interlid region. Once drained, disconnect the drain line.

9. Install at least 10 inner lid bolts equally spaced on the bolt circle to hand tight.
10. Continue raising the cask from the pool while spraying the external cask surfaces with clean water to minimize surface contamination levels.
11. Move the cask to the cask decontamination area, lower the cask to the floor and disengage the lift yoke (or lift beam and inner lid lifting slings if the alternate method of handling the inner lid was used). Remove the lift yoke and crane from the area.
12. Connect a vent line to the vent port quick-disconnect. Direct the free end of the vent line to a radioactive waste handling system capable of handling liquids and gas.
13. Remove the inner lid alignment pins and install the remaining inner lid bolts and torque all of the bolts to the torque value specified in Table 7-1. The bolt torquing sequence is shown on the inner lid.
14. Connect a drain line to the drain port quick-disconnect (located in the inner lid). Remove the vent line from the vent port quick-disconnect.
15. Drain the cask cavity by connecting a nitrogen or helium supply to the vent port quick-disconnect (located in the inner lid). Purge the water from the cask by pressurizing to 35 to 40 psig and hold until all water is removed (observed when no water is coming from the drain line). Turn the nitrogen or helium supply off and disconnect the nitrogen or helium supply line from the vent port. Then, disconnect the drain line from the drain port quick-disconnect.
16. Connect a vacuum pump to the cask cavity via the vent and drain port quick-disconnects in the inner lid. Evacuate the cask cavity until a pressure of 4 mbar is reached. Continue pumping for a minimum of 1 hour after reaching 4 mbar. Valve off vacuum pump from system and using a calibrated vacuum gauge (minimum gauge readability of 2.5 mbar), observe for a pressure rise. If a pressure rise (ΔP) of more than 12 mbar in ten minutes is observed, continue pumping until the pressure does not rise more than 12 mbar in ten minutes. Repeat dryness test until cavity dryness has been verified ($\Delta P < 12$ mbar in 10 minutes). Record test results in the cask loading report.
17. Without allowing air to re-enter the cask cavity, turn off and isolate the vacuum pump. Connect a supply of helium (99.9% minimum purity) to the vent port quick-disconnect and backfill the cask cavity to 0 psig helium pressure.
18. Install the drain and vent port coverplates using new metallic O-rings or inspected Viton O-rings. Torque the bolts to the value indicated in Table 7-1.
19. Perform inner lid O-ring leakage testing as follows:
 - 19a. For the metallic O-ring assembly, connect the leak detector vacuum pump to the inner lid interseal test port and evacuate the air between the O-rings to <1 mbar. Hold the

- vacuum on the interseal for the metallic O-ring assembly region. Using the helium leak detector, verify that any detectable leak rate for metallic O-rings is $\leq 2 \times 10^{-7}$ cm³/sec (helium). The test sensitivity shall be $\leq 1 \times 10^{-7}$ cm³/sec (helium).
- 19b. For Viton O-rings, perform the preshipment leakage rate test to confirm no detected leakage to a test sensitivity of 1×10^{-3} ref cm³/sec by pressurizing the O-ring annulus to 15 (+2, -0) psig and isolating for a minimum of 15 minutes. There shall be no loss in pressure during the test period.
 - 19c. For new replacement Viton O-rings, use a leak detector connected to the interseal test port to verify the total leakage rate is $\leq 9.3 \times 10^{-5}$ cm³/sec (helium)⁽¹⁾ with a minimum test sensitivity of 4.7×10^{-5} cm³/sec (helium).
 20. Install the test port plug for the inner lid interseal test port using a new metallic O-ring and torque the plug to the value specified in Table 7-1.
 - 21a. For the metallic O-ring, connect a vacuum pump to the vent port coverplate interseal test port and evacuate the air between the O-rings to <1 mbar. Hold the vacuum on the interseal for the metallic O-ring assembly region. Using the helium leak detector, verify that any detectable leak rate for metallic O-rings is $\leq 2 \times 10^{-7}$ cm³/sec (helium). The test sensitivity shall be $\leq 1 \times 10^{-7}$ cm³/sec (helium).
 - 21b. For Viton O-rings⁽¹⁾, perform the preshipment leakage rate test to confirm no detected leakage to a test sensitivity of 1×10^{-3} ref cm³/sec by pressurizing the O-ring annulus to 15 (+2, -0) psig and isolating for a minimum of 15 minutes. There shall be no loss in pressure during the test period.
 - 21c. For new replacement Viton O-rings, use a leak detector connected to the interseal test port to verify the leakage rate is $\leq 9.3 \times 10^{-5}$ cm³/sec (helium)⁽¹⁾ with a minimum test sensitivity of 4.7×10^{-5} cm³/sec (helium).
 22. Install the test port plug for the vent port coverplate using a new metallic O-ring and torque the plugs to the value specified in Table 7-1.
 23. Repeat Steps 21 and 22 for the drain port coverplate.⁽¹⁾
 24. Drain residual water from the pressure port, ensuring that the pressure port is clear to also allow water to drain from the interlid region.

(1) For new Viton O-rings, the combined leakage rates are for the inner O-ring of the lid, inner O-ring of the vent port cover plate and inner O-ring of the drain port cover plate, which are part of the containment boundary. The combined measured leakage rate from all three Viton O-rings must be less than or equal to $\leq 9.3 \times 10^{-5}$ cm³/sec (helium) in accordance with 10 CFR 71.51.

25. Install the transport pressure port cover on the pressure port. Torque the port cover bolts to the value specified in Table 7-1.
26. Perform a functional leak test on the pressure port cover by removing the O-ring test plug and using a test fixture, pressurize the annulus between the pressure port cover O-rings to 15 psig and isolate. During a 10-minute test period, there shall be no loss in pressure during the test period.
27. Install the pressure port cover interseal test port plug and O-ring and torque the plug to the value specified in Table 7-1.
28. For the metallic outer lid O-ring assembly, remove the O-ring, clean the O-ring seating surface and groove, and install a new metallic O-ring. For Viton O-ring assemblies, inspect the O-ring and replace if damaged.
29. Install outer lid and align vent pins.
30. Attach the outer lid lifting device to the outer lid and overhead crane. Install the outer lid using the alignment pins to assist in proper seating. Remove the outer lid alignment pins. Install the outer lid bolts and torque to the value specified in Table 7-1. The bolt torquing sequence is shown on the outer lid.
31. Attach a supply of air or helium to the interlid port quick-disconnect. Backfill the interlid volume to 15 psig air or helium and hold for 10 minutes. There shall be no pressure loss during the test period. Disconnect air or helium supply.
32. Install the interlid port cover using new metallic O-rings. Torque the interlid port cover bolts to the value specified in Table 7-1.
33. Remove the test plug from the interlid port cover and, using the O-ring test fixture, pressurize the O-ring annulus to 15 psig with air or helium. Isolate the annulus and hold for 10 minutes. No loss of pressure is permitted during the test period.
34. Remove the air or helium supply and vent the annulus pressure. Replace the metallic O-ring on the interlid port cover test plug, install the test plug and torque it to the value specified in Table 7-1.
35. Perform final external decontamination and perform survey to verify acceptable level of removable contamination to ensure compliance with 49 CFR 173.443. Perform final radiation survey. Record the survey results.
36. Perform final visual inspection to verify assembly of the NAC-STC in accordance with the Certificate of Compliance. Verify that the loading documentation has been appropriately completed and signed off.

7.1.3.2 Loading Canistered Fuel or GTCC Waste

Canistered fuel or canistered GTCC waste is loaded into the NAC-STC using a transfer cask. This procedure assumes that the canister has been previously loaded, drained, vacuum dried, backfilled with helium and welded closed. The canister may have been retrieved from dry storage, or it may have been loaded and sealed immediately prior to loading in the NAC-STC. This procedure assumes the sealed canister conforms to the design basis of the NAC-STC and that the canister is already in the transfer cask.

1. Attach the transfer cask yoke to the cask handling crane hook.
2. Engage the transfer cask yoke to the trunnions of the transfer cask.
3. Raise the transfer cask over the NAC-STC cask and lower it until it rests on the transfer cask adapter plate. Remove and store the transfer cask lifting yoke. Remove the transfer cask shield door stops.
4. Attach the two (2) canister 3-legged lifting sling sets to the hoist rings in the canister lid. Attach the opposite end of the slings to the crane hook.
Note: Alternative canister lifting systems may be utilized.
5. Attach the hydraulic system to the operating cylinders on the transfer cask adapter plate.
6. Raise the canister just enough to take the canister weight off of the transfer cask bottom doors.
7. Open the transfer cask shield doors.
8. Lower the canister into the NAC-STC cask. Exercise caution to avoid contact with the interior cavity wall.
9. Disconnect and remove the canister lifting sling from the crane hook and lower it onto the top of the canister.
10. Close the transfer cask bottom doors and install the door stops.
11. Retrieve the transfer cask lifting yoke and engage the transfer cask trunnions. Lift the transfer cask from the transfer cask adapter plate. Store the transfer cask and transfer cask lifting yoke in the designated locations.
12. Install the NAC-MPC canister top spacer (Yankee-MPC canister only).
13. Retrieve the cask adapter plate lifting sling and attach it to the transfer cask adapter plate.
14. Remove the transfer cask adapter plate and store it in the designated location. Using the appropriate lifting sling, remove the adapter ring and bolts. Install the inner lid alignment pins.

15. Remove the inner lid O-rings and clean inner lid O-ring groove surfaces. Replace the metallic O-rings on the inner lid, carefully inspecting the new O-rings for damage prior to installation. Secure the O-rings in the groove using the O-ring clips and screws.
16. Attach the inner lid lifting slings to an auxiliary crane hook, lift the inner lid and place it on the cask using the inner lid alignment pins to assist in proper lid seating and orientation. Visually verify proper lid position.
17. Disconnect the lid lifting device from the crane hook and remove it from the inner lid.
18. Install at least 10 inner lid bolts equally spaced on the bolt circle to hand tight. Remove the inner lid alignment pins.
19. Install the remaining inner lid bolts and torque all of the bolts to the torque value specified in Table 7-1. The bolt torquing sequence is shown on the inner lid.
20. Remove the metallic O-rings in the drain port coverplate, and clean and inspect the O-ring groove. Install new metallic O-rings and install the coverplate. Torque the coverplate bolts to the value specified in Table 7-1.
21. Connect the vacuum pump to the cask vent port and evacuate the cask cavity to a stable vacuum pressure of 4 mbar (approximately 3 mm of Hg). Without allowing air to re-enter the cask, backfill the cavity with helium (99.9% minimum purity) to 0 psig. Disconnect the helium supply.
22. Remove the metallic O-rings in the vent port coverplate and clean and inspect the O-ring groove. Install new metallic O-rings in the vent port coverplate and install the coverplate. Torque the coverplate bolts to the value specified in Table 7-1.
23. Connect the leak detector to the inner lid interseal test port and evacuate the air between the metallic O-rings until a pressure of <1 mbar is reached. Using the helium leak detector, verify that any detectable leak rate is $\leq 2 \times 10^{-7} \text{ cm}^3/\text{sec}$ (helium). The test sensitivity shall be $\leq 1 \times 10^{-7} \text{ cm}^3/\text{sec}$ (helium).
24. Install the test port plug for the inner lid interseal test port using a new metallic O-ring and torque the plug to the value specified in Table 7-1.
25. Connect the leak detector to the vent port coverplate interseal test port. Evacuate the interseal volume until a pressure of <1 mbar is reached. Using the helium leak detector, verify that any detectable leak rate is $\leq 2 \times 10^{-7} \text{ cm}^3/\text{sec}$ (helium). The test sensitivity shall be $\leq 1 \times 10^{-7} \text{ cm}^3/\text{sec}$ (helium).
26. Install the test port plug for the vent port coverplate using a new metallic O-ring and torque the plug to the value specified in Table 7-1.
27. Repeat Steps 25 and 26 for the drain port coverplate test port.
28. Remove the outer lid metallic O-ring. Clean the outer lid O-ring seating surface and groove. Install a new metallic outer lid O-ring. Install the outer lid alignment pins.

29. Attach the outer lid lifting device to the outer lid and overhead crane. Install the outer lid using the alignment pins to assist in proper seating. Remove the outer lid alignment pins. Install the outer lid bolts and torque to the value specified in Table 7-1. The bolt torquing sequence is shown on the outer lid.
30. Attach a supply of air or helium to the interlid port quick-disconnect. Backfill the interlid volume to 15 psig air or helium and hold for 10 minutes. No loss of pressure is permitted during the 10-minute test period. Disconnect air or helium supply.
31. Install the transport interlid port cover in the interlid port using new O-rings. Torque the interlid port cover bolts to the value specified in Table 7-1.
32. Remove the O-ring test plug from the interlid port cover and, using the O-ring test fixture, pressurize the O-ring annulus to 15 psig with air or helium. Isolate the annulus and hold for 10 minutes. No loss of pressure is permitted during the test period.
33. Vent the annulus pressure, remove the air or helium supply, replace the metallic O-ring on the interlid port cover test plug and install the test plug. Torque the plug to the value specified in Table 7-1.
34. Perform final external decontamination and perform survey to verify acceptable level of removable contamination to ensure compliance with 49 CFR 173.443. Perform final radiation survey. Record the survey results in the cask loading report.
35. Perform final visual inspection to verify assembly of the NAC-STC in accordance with the Certificate of Compliance. Verify that the loading procedure and checklist are appropriately completed and signed off.

7.2 Preparation for Transport

Perform the procedures of either Section 7.2.1 or 7.2.2, whichever is appropriate. Section 7.2.1 addresses preparation for transport without interim storage after loading the cask either with directly loaded fuel or with a previously loaded canister. Section 7.2.2 addresses transport following long-term storage of directly loaded fuel. Transport following long-term storage requires the verification of containment by leak testing the containment boundary formed by the outer O-rings of the inner lid and port covers and the O-ring test ports.

7.2.1 Preparation for Transport (Immediately After Loading)

1. Engage the lift beam to the cask lifting trunnions and move the cask to the cask loading area.
2. Load the cask onto the transport vehicle by gently lowering the rotation trunnion recesses into the rear support. Rotate the cask to horizontal by moving the overhead crane in the direction of the front support. Maintain the crane cables vertical over the lifting trunnions.
3. Using a lifting sling, place the tiedown assembly over the cask upper forging between the top neutron shield plate and front trunnions. Install the front tiedown bolts and lock washers to each side of the front support.
4. Complete a Health Physics removable contamination survey of the cask to ensure compliance with 49 CFR 173.443. Complete a Health Physics radiation survey of the entire package to ensure compliance with 49 CFR 173.441.
5. Using the designated lifting slings and a crane of appropriate capacity, install the top impact limiter. Install the impact limiter retaining rods into each hole and torque to the value specified in Table 7-1. Install the impact limiter attachment nuts and torque to the value specified in Table 7-1. Install the impact limiter jam nuts and torque to the value specified in Table 7-1. Install the impact limiter lock wires. Repeat the operation for the bottom impact limiter installation.

Note: Balsa impact limiters shall be used for transport of the Connecticut Yankee fuel or GTCC waste canisters. The balsa impact limiters may also be used for transport of directly loaded fuel and for canisters containing Yankee fuel or GTCC waste. Redwood impact limiters may be used for transport of directly loaded fuel and for canisters containing MPC-Yankee fuel or GTCC waste.

6. Install security seals through holes provided in the upper impact limiter and one of the lifting trunnions; and through holes provided in all three bolts in the interlid port cover and the pressure port cover. Record the security seal identification numbers in the cask loading report.
7. Install the personnel barrier/enclosure and torque all attachment bolts to the prescribed torque value. Install padlocks on all personnel barrier/enclosure accesses.
8. Complete a Health Physics radiation survey of the entire package to ensure compliance with 49 CFR 173.441.
9. Complete a Health Physics removable contamination survey of the transport vehicle to ensure compliance with 49 CFR 173.443.
10. Determine the transport index (TI) corresponding to the maximum dose rate at 1 meter from the cask. Record on the shipping documents.
11. Determine the appropriate Criticality Safety Index (CSI) assigned to the package contents in accordance with the CoC, and indicate the correct CSI on the fissile material labels applied to the package.
12. Apply placards to the transport vehicle in accordance with 49 CFR 172.500 and provide special instructions to the carrier/shipper for an Exclusive Use Shipment.
13. Complete the shipping documentation in accordance with 49 CFR Subchapter C.

7.2.2 Preparation for Transport (After Long-Term Storage)

This procedure applies to the transport of directly loaded fuel that has been in storage in the NAC-STC. Canistered fuel or canistered GTCC waste may not be loaded in the NAC-STC for storage. Canistered fuel or GTCC waste is loaded into the NAC-STC only for transport without interim storage.

Prior to placing a directly loaded cask in long-term storage, the cask cavity is backfilled with 1.0 atmosphere (absolute) of helium (99.9% minimum purity) as the normal coolant for the spent fuel and to provide an inert atmosphere to prevent possible oxidation of the fuel. The inner lid interseal volume between the two inner lid metallic gaskets and the interseal volume between the O-rings in the vent and drain port covers are backfilled with 15 psig of helium (99.9% minimum purity). The interlid volume is pressurized to 100 psig and that pressure is monitored for pressure loss by a pressure transducer installed in the cask upper forging, and closed by a specially equipped port cover filled with a pressure feed-through tube (License Drawing No. 423-807). This overpressure system ensures that in the off-normal event of any leakage of the inner lid or port cover O-rings, the leakage path will be clean helium into the cavity. If, during the storage period, no significant pressure loss is observed in the pressure monitoring volume or

system (normally recorded at a minimum of once every 24 hours during storage), it can be concluded that at the end of the storage period, the cask cavity remains backfilled with helium gas.

Prior to preparing the cask for transport, the pressure transducer wiring has been disconnected.

1. Move cask from extended storage location to a designated work area.
2. Evacuate a sample bottle using a vacuum pump and remove the interlid pressure port cover. Isolate the sample bottle and connect it to the interlid port quick-disconnect and fill it with interlid region atmosphere.
Note: The interlid pressure may be as high as 100 psig. Use caution in collecting the gas sample.
3. Isolate the sample bottle and disconnect it from the interlid port quick-disconnect.
4. Bring the sample bottle to the appropriate facility and analyze the contents of the sample bottle.
5. If krypton-85 is present in the sample bottle, additional radiological precautions may be imposed by Health Physics personnel prior to proceeding with the removal of the outer lid. A determination shall also be made as to whether replacement of the inner lid seals is required. If the gas sample is acceptable, proceed with normal operations.
6. Attach valved venting hose to interlid port quick-disconnect and open valve to vent interlid region.
7. Remove the outer lid bolts and install the outer lid alignment pins and outer lid lifting eye bolts.
8. Attach the outer lid lifting device to the outer lid lifting eye bolts and overhead crane. Remove the outer lid and place it aside in a temporary storage area. Protect the O-ring and O-ring groove of the lid from damage. Remove the outer lid alignment pins.
9. Verify the torque of the inner lid bolts and vent and drain port coverplate bolts by torquing the bolts in accordance with the bolt torque sequence to the values specified in Table 7-1.
10. Remove the drain port coverplate port plug. Connect the leak detector vacuum pump to the drain port coverplate test port and evacuate the helium between the metallic O-rings to a pressure of <1 mbar. Without allowing air to re-enter the interseal region, backfill the drain port coverplate interseal region with helium (99.9% minimum purity) to a pressure of 0 psig.
11. Install the drain port coverplate test plug using a new O-ring and torque to the value specified in Table 7-1.
12. Repeat steps 10 and 11 for the vent port coverplate test plug.

13. Remove the inner lid interseal test port plug and connect a vacuum pump to the inner lid interseal test port quick-disconnect. Evacuate the inner lid interseal volume until a pressure of <1 mbar.
14. Without allowing air to re-enter the interseal volume, backfill the interseal volume with helium (99.9% minimum purity) to 0 psig. Disconnect helium supply.
15. Install the inner lid interseal test port plug with a new metallic O-ring and torque the plug to the value specified in Table 7-1.
16. Clean the outer lid O-ring seating surface and groove surface. Install a new metallic O-ring in the outer lid. Reinstall the outer lid alignment pins.
17. Attach the outer lid lifting device to the outer lid lifting eye bolts and the overhead crane. Install the outer lid and visually verify proper seating. Remove the alignment pins and lifting eye bolts, and install the outer lid bolts and torque to the value specified in Table 7-1. The bolt torquing sequence is shown on the outer lid.
18. Perform an evacuated envelope leakage test on the outer O-rings of the vent and drain port coverplates, the outer O-ring of the inner lid, and the interseal test ports by connecting a vacuum pump and a helium mass spectrometer leak detector connected to the interlid port quick-disconnect. Evacuate the interlid region to a vacuum of <1 mbar.
19. Using the helium leak detector, verify that the leakage rate into the evacuated envelope is $\leq 2 \times 10^{-7}$ cm³/sec (helium) with a minimum leak test sensitivity of $\leq 1 \times 10^{-7}$ cm³/sec.
20. Upon completion of the leak test, backfill the interlid region with helium (99.9% minimum purity) to 0 psig and disconnect the helium supply and leak test equipment.
21. Install the transport interlid port cover using new O-rings and torque the port cover bolts to the value specified in Table 7-1.
22. Remove the interseal port plug, attach the test fixture to the interlid port interseal test hole and perform a functional leak test on the interlid port cover O-rings by pressurizing the O-ring annulus to 15 psig and isolating for a minimum of 10 minutes. There shall be no loss in pressure during the test period. Record completion of an acceptable leakage test on the cask loading report. Upon completion of the test, equalize interseal region pressure with ambient and disconnect the test fixture. Install the interseal port plug and torque to the value specified in Table 7-1.
23. Using the lift yoke, load the cask on the transport vehicle.
24. Using a lifting sling, place the tiedown assembly over the cask upper forging between the top neutron shield plate and front trunnions. Install the front tiedown bolts and lock washers to each side of the front support.

25. Complete a Health Physics removable contamination survey of the entire package to ensure compliance with 49 CFR 173.443.
26. Using the designated lifting slings and a crane of appropriate capacity, install the top impact limiter. Install the impact limiter retaining rods into each hole and torque to the value specified in Table 7-1. Install the impact limiter attachment nuts and torque to the value specified in Table 7-1. Install the impact limiter jam nuts and torque to the value specified in Table 7-1. Install the impact limiter lock wires. Repeat the operation for the bottom impact limiter installation.
27. Install security seals through holes provided in the upper impact limiter and one of the lifting trunnions; and through holes provided in all three bolts in the interlid port cover and the pressure port cover.
28. Install personnel barrier/enclosure and torque all attachment bolts to the prescribed torque value. Install padlocks on all personnel barrier/enclosure accesses.
29. Complete radiation and contamination surveys to ensure compliance with 49 CFR 173.441 and 173.443 requirements.
30. Determine the transport index (TI) corresponding to the maximum dose rate at 1 meter from the cask. Record on the shipping documents.
31. Determine the appropriate Criticality Safety Index (CSI) assigned to the package contents in accordance with the CoC, and indicate the correct CSI on the fissile material labels applied to the package.
32. Apply placards to the transport vehicle in accordance with 49 CFR 172.500.
33. Complete the shipping documentation in accordance with 49 CFR Subchapter C and provide special instructions to the carrier/shipper for an Exclusive Use Shipment.

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7.3 Outline of Procedures for Unloading the Cask

This section presents the procedures to be followed for unloading the cask following transport of directly loaded fuel, canistered fuel or canistered GTCC waste.

7.3.1 Receiving Inspection

1. Perform radiation and removable contamination surveys in accordance with 10 CFR 20.1906, 49 CFR 173.441 and 173.443 requirements.
2. Remove the personnel barrier/enclosure and complete radiation and removable contamination surveys at the cask surfaces.
3. Visually inspect the NAC-STC while secured to the transport vehicle in the horizontal orientation for any signs of damage and record any damage. Verify that the tamper-indicating seals are in place and verify their numbers.
4. Secure the transport vehicle. Attach slings to the top impact limiter lifting points, remove impact limiter lock wires, jam nuts, attachment nuts and retaining rods, and remove the impact limiter. Store the impact limiter upright. Repeat the operation to remove the bottom impact limiter. Complete radiation and removable contamination surveys for exposed cask surfaces.
5. Release the tiedown assembly from the front support by removing the front tiedown bolts and lock washers.
6. Attach a sling to the tiedown assembly lifting eyes and remove the tiedown assembly from the transport vehicle.
7. Attach the cask lifting yoke to a crane hook with the appropriate load rating. Engage the two yoke arms with the lifting trunnions at the top end of the cask. Rotate/lift the cask to the vertical orientation and raise the cask off of the rear support structure of the transport vehicle. Place the cask in the vertical orientation in a decontamination area or other location identified by the user.
8. Wash any road dust and dirt off of the cask and decontaminate cask exterior, as required by contamination survey results.

7.3.2 Preparation of the NAC-STC Cask for Unloading

The NAC-STC may contain fuel directly loaded into a basket within the cask, or a sealed transportable storage canister containing spent fuel assemblies, Reconfigured Fuel Assemblies, Recaged Fuel Assemblies, four fuel assemblies loaded in Damaged Fuel Cans, or GTCC waste.

Unloading of fuel from the directly loaded cask basket typically takes place under water in the spent fuel pool cask loading area. Canister unloading is performed dry using a transfer cask. Canister unloading will take place in the cask receiving area, or other location identified by the user.

7.3.2.1 Preparation for Unloading the NAC-STC Cask (Directly Loaded Fuel Configuration)

1. Verify that excessive pressure does not exist in the interlid region by removing the interlid port cover and attaching a pressure test fixture to the interlid port quick-disconnect that will allow the monitoring of the cask interlid region for any pressure buildup that may have occurred during transport. If a positive pressure exists, connect a vent/drain line to the interlid quick-disconnect and vent the pressure to the off-gas system.
2. Remove the outer lid bolts and install the outer lid alignment pins and outer lid lifting eye bolts.
3. Attach the outer lid lifting device to the outer lid lifting eye bolts and the overhead crane. Remove the outer lid and place it aside in a temporary storage area. Protect the O-ring and the O-ring groove of the lid from damage.
4. Remove the port coverplates from the drain and vent ports in the inner lid with caution. Attach a pressure test fixture to the vent port that will allow the monitoring of the cask cavity for any pressure buildup that may have occurred during storage or transport. If a positive pressure exists, vent the pressure to the off-gas system.
5. Connect the cask cooldown system to the drain and vent quick-disconnects. The cask cooldown piping and controls schematic is shown in Figure 7.3-1.
6. To facilitate cooldown and to minimize thermal effects to the cask and its contents, slowly (8 – 10 gpm) fill the cask cavity with clean demineralized water (cavity is full when water flows out of the vent port drain line). Circulate water through the cask until the water leaving the vent port drain line is within 50°F of the average spent fuel pool water temperature.
7. Disconnect the fill line from the drain port quick-disconnect in the inner lid (Note: Leave a short drain line attached to the vent port quick-disconnect for continuous venting).
8. Loosen and remove all but 10, approximately equally spaced, inner lid bolts. Leave the 10 remaining inner lid bolts hand tight. Install the inner lid alignment pins at locations marked on the inner lid and the lid lifting eyebolts.

9. Remove the interlid port cover from the top forging. Disengage the vent line from the vent port quick-disconnect.
10. Attach the lifting yoke to a crane hook and engage the yoke arms with the lifting trunnions. Lift the cask and move it over to the cask loading area in the pool.
11. Spray the external surface of the cask with clean demineralized water to minimize external decontamination efforts. Slowly lower the cask into the pool. Just prior to submerging the top forging of the cask, complete the unthreading of the 10 remaining inner lid bolts and remove them.

Note: Use caution when removing these bolts as pressure may rise slightly in the cask during the time since completion of Step 9.

12. Continue lowering the cask until it rests in the cask loading area on the pool floor.
13. Disconnect the lifting yoke from the lifting trunnions and move the yoke so that it will not interfere with fuel movements.
14. Using the inner lid lifting device attached to an auxiliary crane hook, remove the inner lid from the cask.

Note: If the alternate method of handling the cask is being used, slowly raise the lift yoke and the inner lid using the lid alignment pins to guide movement. Move the lift yoke and the inner lid out of the area so that it will not interfere with fuel movements.

15. Place the inner lid aside ensuring that the O-rings and O-ring grooves are protected from damage. Decontaminate, as necessary, and clean all sealing surfaces.

7.3.2.2 Preparation for Unloading the NAC-STC Cask (Canistered Configuration)

1. Verify that excessive pressure does not exist in the interlid region by removing the interlid port cover and attaching a pressure test fixture to the interlid port quick-disconnect that will allow the monitoring of the cask interlid region for any pressure buildup that may have occurred during transport. If a positive pressure exists, connect a vent line to the interlid quick-disconnect and vent the pressure to the off-gas system.
2. Remove the outer lid bolts and install the outer lid alignment pins and outer lid lifting eye bolts.
3. Attach the outer lid lifting device to the outer lid lifting eye bolts and the overhead crane. Remove the outer lid and place it aside in a temporary storage area. Protect the O-ring and the O-ring groove of the lid from damage. Remove the outer lid alignment pins.

4. Remove the port coverplates from the drain and vent ports in the inner lid with caution. Attach a pressure test fixture to the vent port that will allow the monitoring of the cask cavity for any pressure buildup that may have occurred during transport. If a positive pressure exists, vent the pressure to the off-gas system.
5. Loosen and remove all inner lid bolts. Install the inner lid alignment pins at locations marked on the inner lid and the lid lifting hoist rings.
6. Using the inner lid lifting slings, attached to a suitable crane, remove the inner lid from the cask. Remove the inner lid alignment pins.
7. Place the inner lid aside ensuring that the O-rings and O-ring grooves are protected from damage. Decontaminate, as necessary, and clean all sealing surfaces.
8. If present, remove the top spacer from the NAC-STC cask cavity.
9. Install the adapter ring on the NAC-STC and torque the three captive bolts to the torque specified in Table 7-1.
10. Install the transfer cask adapter plate on the top surface of the cask and remove the handling slings.

7.3.3 Unloading the NAC-STC Cask

The NAC-STC may contain either fuel directly loaded in the cask basket, or a welded transportable storage canister. The procedures for unloading the directly loaded fuel or canisters are presented in the following.

7.3.3.1 Unloading Directly Loaded (Uncanistered) Fuel

1. Using approved fuel identification and handling procedures, withdraw one fuel assembly from the basket and deposit it in the proper storage rack location. Be careful not to contact any of the sealing surfaces on the top forging or the inner lid alignment pins.
2. Record and document the fuel movement from the cask to the fuel rack.
3. Repeat steps 1 and 2 until all fuel assemblies have been removed from the cask.
4. Attach the inner lid lifting slings to a crane hook, lift the inner lid and place it on the cask using the alignment pins to assist in proper seating. Visually verify proper lid position.

Note: O-ring seals on the lids, port coverplates and test plugs do not require replacement for an empty packaging shipment.
5. Disconnect the lid-lifting sling from the crane hook.

6. Attach the lifting yoke to the crane hook, lower to lifting position and engage lifting arms to lifting trunnions. Slowly lift the cask out of the pool until the top of the cask is slightly above the pool water level.

Note: As an alternative method, the cask and inner lid may be handled simultaneously.

In the event that this method is chosen, instead of performing steps 4, 5 and 6, attach the lifting yoke to a crane hook and the inner lid to the lift yoke. Lower the lid and engage to the cask using the lid alignment pins. Engage lifting arms to lifting trunnions. Slowly lift the cask out of the pool until the top of the cask is slightly above the pool water level.

7. Attach a drain line to the quick-disconnect in the interlid port (located in the top forging) and allow the water to drain from the interlid region.
8. Install at least four inner lid bolts approximately equally spaced on the bolt circle to hand tight. Remove the inner lid alignment pins.
9. Move the NAC-STC cask to the cask decontamination area and disengage the lift yoke or lift beam and inner lid lifting slings if the alternate method of handling the inner lid was used. Remove the inner lid lifting eye bolts.
10. Move the cask lifting equipment away from the cask work area.
11. Install the remaining inner lid bolts and torque all of the inner lid bolts to the value specified in Table 7-1 in accordance with the bolt torquing sequence shown on the inner lid.
12. Disconnect the drain line from the quick-disconnect in the interlid port.
13. Connect a drain line to the drain port quick-disconnect and a regulated air fill line to the vent port quick-disconnect.
14. Purge the water from the cask by pressurizing to 35 to 40 psig and hold until all water is removed (observed when no water is coming from the drain line). Adjust final internal cavity pressure to 0 psig.
15. Remove the lines from the drain and the vent port quick-disconnects.
16. Install the port coverplates over the vent and drain ports in the inner lid. Torque the coverplate bolts to the value specified in Table 7-1.
17. Decontaminate the surfaces of the inner lid and the inner surfaces of the top forging.
18. Install the outer lid alignment pins. Using the outer lid lifting device, install the outer lid using the alignment pins to assist in proper seating. Remove the lid lifting device, lid lifting eyebolts, and the outer lid alignment pins.
19. Install the outer lid bolts and torque them to the value specified in Table 7-1, using the bolt torquing sequence shown on the outer lid.
20. Install the interlid port cover and torque the bolts to the value specified in Table 7-1.

7.3.3.2 Unloading Canistered Fuel or Canistered GTCC Waste

Canistered fuel or GTCC waste is unloaded from the NAC-STC using a transfer cask. The transfer cask could be used to transfer the loaded canister to a work station where the canister could be opened, or to transfer it to another storage or disposal overpack.

1. Install the lift hoist rings in the canister lid.
Note: The canister lid may be thermally hot.
2. Attach the canister lifting sling to the hoist rings in the canister lid. Position the sling so that the free end of the sling can be engaged by the cask handling crane hook.
3. Attach the transfer cask lifting yoke to the cask handling crane hook. Engage the yoke to the lifting trunnions of the transfer cask.
4. Lift the transfer cask and move it over the NAC-STC cask. Lower the transfer cask to engage the transfer cask adapter plate. Once the transfer cask is fully seated, remove the transfer cask lifting yoke and store it in the designated location.

Note: O-ring seals on the lids, port coverplates and test plugs do not require replacement for an empty packaging shipment.

5. Remove the shield door stops, connect the hydraulic operating system and open the transfer cask bottom doors.
6. Using tag lines, lift the canister lifting slings through the transfer cask and attach them to the crane hook.
Note: Alternative canister handling systems may be used.
7. Raise the canister into the transfer cask just far enough to allow the transfer cask bottom doors to close. Use caution to minimize the contact between the canister and the cavity walls of the NAC-STC and of the transfer cask.
8. Close the bottom doors and install the door stops.
9. Carefully lower the canister until it rests on the transfer cask bottom doors. Disengage the canister lifting sling from the crane hook.
10. Retrieve the transfer cask lifting yoke and attach it to the transfer cask trunnions. Lift the transfer cask from the NAC-STC cask and move it to its intended destination.
11. Attach the transfer cask adapter plate-lifting slings and disconnect the hydraulic operating system.
12. Using the crane, lift the transfer cask adapter plate from the top of the cask. Move the transfer cask adapter plate to the designated storage location.
13. Detorque the three bolts and remove the adapter ring.

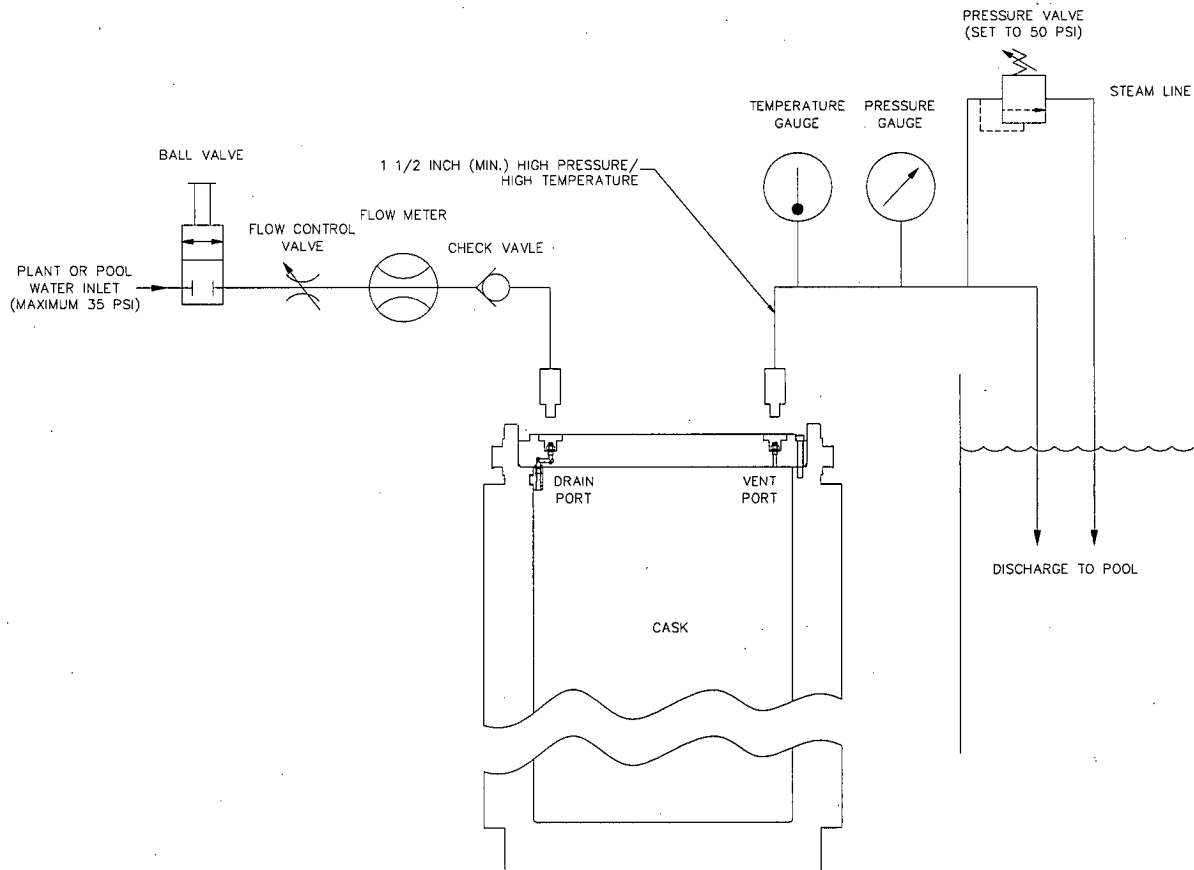
14. Install the inner lid alignment pins.
15. Attach the inner lid lifting fixture to the inner lid and engage the lifting fixture to the auxiliary crane. Install the inner lid in the NAC-STC using the alignment pins to assist in proper seating.
16. Disconnect the lifting fixture and remove the guide pins.
17. Install and torque the inner lid bolts to the values specified in Table 7-1 using the bolt torquing sequence shown on the inner lid.
18. Install the port coverplates over the vent and drain ports in the inner lid. Torque the coverplate bolts to the values specified in Table 7-1.
19. Decontaminate the surfaces of the inner lid and the inner surfaces of the top forging.
20. Install the outer lid alignment pins. Using the outer lid lifting device, install the outer lid using the alignment pins to assist in proper seating. Remove the lid lifting device, lid lifting eyebolts, and the outer lid alignment pins.
21. Install the outer lid bolts and torque them to the value specified in Table 7-1 using the bolt torquing sequence shown on the outer lid.
22. Install the interlid port cover and torque the bolts to the value specified in Table 7-1.

7.3.4 Preparation of Empty Cask for Transport

1. Decontaminate all surfaces of the cask to acceptable release limits as defined in 49 CFR 173.
2. Attach the lifting yoke to a crane hook and engage the yoke arms with the lifting trunnions. Lift the cask onto the transport vehicle and lower to the horizontal position.
3. Using a lifting sling, place the tiedown assembly over the cask upper forging between the top neutron shield plate and front trunnions. Install the front tiedown bolts and lock washers to each side of the front support. Torque each of the tiedown bolts.
4. Initiate Health Physics radiation and removable contamination surveys to ensure compliance with 49 CFR 173.441 and 49 CFR 173.443.
5. Using the designated lifting slings and a crane of appropriate capacity, install the top impact limiter. Install the impact limiter retaining rods into each hole and torque to the value specified in Table 7-1. Install the impact limiter attachment nuts and torque to the value specified in Table 7-1. Install the impact limiter jam nuts and torque to the value specified in Table 7-1. Install the impact limiter lock wires. Repeat the operation for the bottom impact limiter installation.
6. Apply labels to the package in accordance with 49 CFR 172.200.

7. Install the personnel barrier/enclosure and torque all attachment bolts to the prescribed torque value. Install padlocks on all personnel barrier/enclosure accesses.
8. Complete the Health Physics radiation and removable contamination surveys to ensure compliance with 49 CFR 173 requirements.
9. Complete the shipping documents.
10. Apply placards, if required, to the transport vehicle in accordance with 49 CFR 172.500.

Figure 7.3-1 Cask Cooldown Piping and Controls Schematic



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7.4 Leak Test Requirements

This section provides the leak testing procedures used to perform the Containment System Verification Leak Tests for the NAC-STC containment boundary O-ring seals. These tests are required following cask loading operations for transport without interim storage for casks provided with metallic seals and after long-term storage in preparation for transport. For transport of uncanistered spent fuel without interim storage, casks provided with Viton O-rings are leak tested to confirm that the containment system is properly assembled for shipment. The preshipment leakage rate test confirms that there is no detected leakage from any seal to a minimum sensitivity of 1×10^{-3} ref·cm³/sec. Detailed procedures, describing the equipment and the leak test system used to perform the leak tests, are developed for use at the licensee's facilities. The containment boundary conditions, required leak tests and leak test acceptance criteria are provided in Table 4.1-1.

The transport cask may be configured with either metallic O-rings or with Viton O-rings. The two types of O-rings may not be used interchangeably, since each O-ring type requires a different O-ring groove configuration. Consequently, the inner lid, vent and drain port coverplates and outer lid are machined with a square O-ring groove to accept metallic O-rings or are machined with a truncated triangular (dove-tail) groove to accept Viton O-rings.

Viton O-rings may be used only when directly loading spent fuel for transport without interim storage. Metallic O-rings must be used when directly loading spent fuel for an extended period of storage and for canistered contents. Metallic O-rings may be used when directly loading spent fuel for transport without interim storage. The metallic and nonmetallic O-rings have different allowable leak rates, as specified in the procedures.

7.4.1 Containment System Verification Leak Test Procedures

As described in Chapter 4, the NAC-STC primary containment boundary is designed and tested to assure that there is no leakage under any of the normal conditions of transport or accident conditions that exceeds the allowable value determined in accordance with 10 CFR 71.51. This leakage rate is verified prior to transport by the performance of leak tests on the containment boundary filled with metallic O-rings to ensure that the leakage rate is less than 2×10^{-7} cm³/sec (helium). For casks provided with Viton O-rings, the Containment System Verification Leak Test is performed annually or after replacement of a Viton O-ring, and the cumulative leak rate is less than 9.3×10^{-5} cm³/sec (helium). As described in Section 4.1, the containment boundary is

defined differently for transport after long-term storage than for loading for transport without interim storage. As described in this section, leak tests are performed in accordance with the requirements of ANSI N14.5-1997.

The leak test requirements and acceptance criteria performed after long-term storage in preparation for transport and performed following cask loading operations for transport without interim storage are described in Sections 7.4.2 and 7.4.3, respectively. The generic procedures used to perform leak testing are incorporated in the NAC-STC loading procedures in Section 7.2. Detailed procedures, describing the equipment and the leak test system used to perform the leak tests, are developed for use at the licensee's facilities. As noted in Section 7.1, the transportable storage canister will have been loaded, closed and sealed prior to loading into the NAC-STC. The canister is a separate inner container for the transport of damaged fuel.

Section 7.4.4 provides the procedural guidance on corrective actions to be taken in the event a leak test does not meet the acceptance criteria.

7.4.2 Leak Testing for Transport After Long-Term Storage

This section summarizes the leak test method used to demonstrate continued containment of PWR spent fuel prior to transport following an extended period of storage. The containment boundary for this transport condition is defined as Containment Condition A in Section 4.1 and requires the use of metallic O-rings in the containment boundary. In addition to the steel inner lid and port coverplates, the containment boundary is specified as the outer O-rings of the inner lid and of the vent and drain port coverplates and the O-rings of the test port plugs. As specified in the generic loading procedure, the outer lid must be removed to test the inner lid and the vent and drain port coverplates prior to transport.

To conduct the leak test, the inner seal regions (annulus between the O-rings) of the inner lid and the vent and drain port coverplates are evacuated to less than one millibar, and backfilled to 0 psig with 99.9% pure helium, and the test port plugs are reinstalled. The outer lid is reinstalled using a new metallic O-ring. The interlid region (between the inner and outer lids) is evacuated to a vacuum of 1 millibar, or less. After the vacuum condition is reached, a helium leak detector is used to sample the interlid region for helium leakage past the inner lid outer O-ring, the vent and drain port coverplate outer O-rings, and O-ring test port plugs. The allowable leak rate is $\leq 2 \times 10^{-7}$ cm³/sec (helium) with a minimum test sensitivity of $\leq 1 \times 10^{-7}$ cm³/sec (helium). This test method conforms to A5.4 (evacuated envelope) of Appendix A of ANSI N14.5-1997. If

helium leakage is detected exceeding the criteria, corrective action is taken as described in Section 7.4.4.

The outer lid and pressure port are tested using a pressure drop method to confirm the installation of the outer lid and pressure port O-rings. The interlid region is pressurized using the interlid port to 15 psig with air and the pressure is held for 10 minutes. No loss of pressure is permitted during the test period. Following the test, the interlid region pressure is reduced to 0 psig. The interlid port cover is installed and the annulus between the O-rings of the port cover is tested using the same method. This test confirms the installation of the interlid port cover O-rings and conforms to test method A.5.1 (gas pressure drop) of Appendix A of ANSI N14.5-1997.

7.4.3 Leak Testing for Transport After Loading without Interim Storage

This section summarizes the leak tests required to demonstrate containment of directly loaded PWR spent fuel without interim storage, or for sealed transportable storage canisters containing spent fuel or GTCC waste. The containment boundary for these transport conditions is defined as Containment Condition B in Section 4.1. In addition to the steel inner lid and port coverplates, the containment boundary is specified as the inner O-rings of the inner lid and of the vent and drain port coverplates. The inner lid O-ring and vent and drain port coverplate O-rings are leak tested using the evacuated envelope method (test description A5.4 of Appendix A of ANSI N14.5-1997) with a vacuum in the annulus between the O-rings. The containment boundary O-rings for fuel directly loaded for transport without interim storage may be either metallic or Viton. The containment boundary O-rings for canistered fuel or GTCC waste are required to be metallic O-rings. The leak detector is used to detect helium in the annulus between the O-rings. The allowable leakage rate for each metallic O-ring defined as the containment boundary is $\leq 2 \times 10^{-7}$ cm³/sec (helium) with a minimum test sensitivity of $\leq 1 \times 10^{-7}$ cm³/sec (helium). The allowable cumulative leakage rate for all Viton O-rings defined as the containment boundary is $\leq 9.3 \times 10^{-5}$ cm³/sec (helium) with a minimum test sensitivity of $\leq 4.7 \times 10^{-5}$ cm³/sec (helium). For leak testing prior to transport of a NAC-STC cask with reusable Viton O-rings, a preshipment leakage test is performed to a minimum test sensitivity of 1×10^{-3} ref-cm³/sec. The higher sensitivity test for the Viton O-rings is performed during annual maintenance testing or when the Viton O-rings or other containment components are replaced.

As the metallic O-rings are replaced for each loaded transport canister, the higher sensitivity test is always required. The series of helium leak tests described confirms that the allowable leak

rates are satisfied for the O-rings used in the containment boundary for Containment Condition B. Section 7.4.4 provides the procedural guidance on corrective actions to be taken in the event a leak test does not meet the acceptance criteria.

Following completion of the inner lid and vent and drain port coverplate leak tests, the outer lid and pressure port are tested using a pressure drop method to confirm the installation of the outer lid and pressure port O-rings. The interlid region is pressurized using the interlid port to 15 psig with air and the pressure is held for a minimum of 10 minutes. No loss of pressure is permitted during the test period. Following the test, the interlid region pressure is reduced to 0 psig. The interlid port cover is installed and the annulus between the O-rings of the port cover is tested using the same method. This test confirms the installation of the interlid port cover O-rings. These components form an additional barrier against the release of radioactive material, but are not a containment boundary.

7.4.4 Corrective Action

If a specific component containing an O-ring fails to meet the leak test acceptance criteria for that component, the component is removed and the O-ring removed. The O-ring groove is cleaned and visually inspected to ensure proper cleanliness and surface condition. A new O-ring of the appropriate type (i.e., metallic or Viton) is installed. The removed component is reinstalled and the bolts torqued to the appropriate torque value. The component is then retested in accordance with the applicable test procedure and acceptance criteria.

For the replacement of the inner lid O-ring either immediately after loading, or after extended storage for the directly loaded configuration, it will be necessary to return the cask to the spent fuel pool to remove the inner lid and allow access for inner lid O-ring replacement. For placement of the cask in the fuel pool following extended storage, the procedures for cask unloading (Section 7.3.3) are utilized to prepare and cool down the cask prior to placement in the pool. At cask storage facilities having appropriate dry transfer or hot cell facilities, the inner lid O-ring can be replaced without placement of the cask in a fuel pool for shielding purposes. Prior to removal of the inner lid, a gas sample should be taken at the vent port to verify the condition in the cavity environment. If there are indications that fuel has failed during the storage period, care should be exercised in both flooding the cask and in removing the inner lid.

In the canistered configuration, the NAC-STC inner lid metallic O-rings may be replaced without returning to the pool since the canister confines and shields the spent fuel or GTCC waste.

7.5 Railcar Design and Certification Requirements

The NAC-STC dual-purpose packaging is designed for transport by rail, heavy-haul vehicle, and barge/ship. The NAC-STC intermodal transport skid and tie-down systems will be designed in accordance with the U.S. Department of Transportation (DOT) regulations applicable to the mode of transport to be utilized. The design requirements for the railcar and tie-down components are defined in the following sections.

7.5.1 Railcar and Tie-Down Design Requirements

The railcar and package tie-down system to be used for transporting the NAC-STC will be designed in accordance with the requirements of the Association of American Railroads (AAR) Manual of Standards and Recommended Practices, Section C, Part II, M-1001, and the Field Manual of the AAR Interchange Rules, Rule Number 88.

7.5.2 Railcar Tie-Down Design Loadings

The railcar tie-down system for the NAC-STC package will be designed to withstand the following transport loads acting simultaneously without generating stresses exceeding the tie-down material yield strength:

- Longitudinal: 7.5 g
- Vertical: 4 g
- Lateral: 1.8 g

7.5.3 Railcar and Tie-Down Certification

The NAC-STC railcar and tie-down system design will be submitted to the AAR Mechanical Division for certification and approval in accordance with the AAR rules and requirements.

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7.6 Procedures for Loading and Unloading the Transportable Storage Canister

This section generally describes the procedures for loading and closing, and for opening and unloading, the Transportable Storage Canister. This description is provided for information only since it is intended that the canister be closed, and opened if necessary, using the transfer cask. As described in this SAR, the canister is designed to be transported in the NAC-STC, or stored in the NAC-MPC Storage System for an extended period. The NAC-MPC storage system is described in the NAC-MPC SAR, Docket No. 72-1025. The NAC-MPC FSAR contains more detailed procedures for loading and unloading fuel in the Transportable Storage Canister. Loading of GTCC waste must meet the modified generic criteria as described below.

The transfer cask is primarily a lifting device used to move the canister assembly, and to provide biological shielding when it contains a loaded canister. The transfer cask is used for the vertical transfer of the canister between work stations and the storage or transport casks.

The principal closing and sealing operations are the draining, drying, and helium backfilling of the canister, and sealing it by welding the shield lid, penetration port covers, and structural lid in place. The principal opening operations are cutting the lid welds or canister side wall to remove the structural lid, opening the vent port to sample the canister atmosphere, cutting the shield lid welds to remove the shield lid, and removing the spent fuel or GTCC waste.

7.6.1 Loading and Closing the Transportable Storage Canister

This procedure assumes that the canister and the transfer cask are positioned in the decontamination area or other suitable work station. The staging area should be within the handling "footprint" of the cask handling crane.

1. Visually inspect the Transportable Storage Canister (canister) to ensure that it is clean and free of debris.
2. Place the canister in the transfer cask.
3. Place the transfer cask and canister into the fuel loading pool.
4. Load the previously designated spent fuel assemblies or the GTCC waste into the canister.
Note: Spent fuel and GTCC waste must be loaded in separate canisters.
5. Place the shield lid on top of the loaded canister.

6. Remove the transfer cask with the loaded canister from the fuel loading pool.
7. Insert the drain tube assembly and remove approximately 50 gallons of water from the Yankee-MPC canister or 65 gallons from the CY-MPC canister.
8. Weld the shield lid in place and verify the adequacy of welds with liquid penetrant examinations. Record results of examinations as required.
9. Pressurize the Yankee-MPC canister to 15 psig with air, nitrogen, or helium and hold the pressure for 10 minutes. For the CY-MPC canister, a pressure of 20 psig is to be used and held for 10 minutes.
10. Release the pressure.
11. Drain the remaining water from the canister.
12. Vacuum dry the canister.
13. For fuel canisters, evacuate to $\leq 3\text{mm Hg}$. For GTCC waste canisters, evacuate to $\leq 10\text{mm Hg}$.
14. Backfill canister with helium to a pressure of one atmosphere.
15. For fuel canister only, evacuate canister to a vacuum of $\leq 3\text{mm Hg}$.
16. For fuel canister only, backfill canister with helium to a pressure of one atmosphere.
17. Remove vacuum drying and helium equipment from the canister vent and drain.
18. Install vent and drain port covers and weld into place. Verify the adequacy of welds with liquid penetrant examinations. Record results of examinations as required.
19. Install the leak test lid and connect the leak detector system to the test lid. Evacuate the test lid volume and verify the leaktightness of the shield lid welds.
20. Install the structural lid in place.
21. Weld the structural lid in place and verify the adequacy of welds with liquid penetrant examinations. Record results of examinations as required.
22. Perform a smear survey of the accessible areas of the canister to ensure that the surface contamination is within limits.
23. Install the transfer cask retaining ring.
24. Decontaminate the external surface of the transfer cask to the limits established for the site.

At this point, the loaded canister may be transferred into the NAC-STC in accordance with the procedures presented in Section 7.1, contingent upon the loaded contents meeting the requirements for the authorized contents for the transport cask system, as described in the cask Certificate of Compliance.

7.6.2 Opening and Unloading the Transportable Storage Canister

Circumstances could arise that dictate the opening of a previously loaded canister and the removal of the stored spent fuel or GTCC waste. This section describes the basic operations needed to open the sealed canister. It is assumed that the canister is positioned in the transfer cask and that the transfer cask is in a suitable work station. The work station must provide for control of airborne radioactive material and gases that could potentially be released from the open canister. It is not intended that the canister be opened while it is in the NAC-STC cask. The principal mechanical operations are the cutting of the closure welds, filling with water, and removing the spent fuel or GTCC waste. Supplemental shielding is used as required.

1. Place the canister in the transfer cask.
2. Survey the top of the canister to establish the radiation level and contamination level at the structural lid.
3. Set up the weld cutting equipment to cut the structural lid weld.
4. Operate the cutting equipment to cut the structural lid weld.
5. Remove the cutting equipment and using a three-legged sling, remove the structural lid.
6. Cut the weld joining the vent port cover to the shield lid.
7. Remove the vent port cover.
8. Sample the canister cover gas and vent any pressure in the canister to a radioactive waste handling system.
9. Cut the weld joining the drain port cover to the shield lid and remove the drain.
10. Attach a nitrogen gas line to the drain port quick-disconnect and a discharge line from the vent port quick-disconnect to an off-gas handling system.
11. Continue to flow nitrogen through the line until there is no evidence of fission gas activity in the discharge line (or 10 minutes minimum).
12. Attach a source of clean water with a minimum temperature of 70°F and a maximum supply pressure of 15 psig to the drain port quick-disconnect. Replace the vent port quick-disconnect with a straight-through fitting fitted with a Viton O-ring, and attach the discharge water line. Slowly start the flow of clean water to establish a flow rate of 5 (+ 3, - 0) gpm.
13. Continue to flow water through the canister until the exit water temperature stabilizes to a temperature below 200°F.
14. Stop the flow of water and remove the connection to the drain line.
15. Set up the weld cutting equipment to cut the shield lid weld.

16. Remove approximately 50 gallons of water from the Yankee-MPC canister or 65 gallons of water from the CY-MPC canister.
17. Using a hydrogen gas detector, check the vent port for hydrogen gas. Purge the hydrogen gas if the concentration of hydrogen gas exceeds 2.4%.
18. Cut the shield lid weld. Attach the shield lid lifting sling.
19. Attach the clean water line to the transfer cask.
20. Retrieve the transfer cask lifting yoke and engage the transfer cask lifting trunnions.
21. Move the transfer cask over the pool and lower the bottom of the transfer cask to the surface. Start the flow of clean water to the transfer cask annulus. Continue to lower the transfer cask, as the annulus fills with clean water, until the top of the transfer cask is about 4 inches above the pool surface. Hold the cask in this position until clean water fills the top of the transfer cask.
22. Lower the transfer cask to the bottom of the cask loading area and remove the lifting yoke.
23. Attach the shield lid lifting sling to the crane hook.
24. Slowly lift the shield lid. Move the shield lid to one side after it is raised clear of the transfer cask.
25. Visually inspect the spent fuel or GTCC waste.

At this point, the spent fuel or GTCC waste can be removed from the canister.

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8.0 ACCEPTANCE TESTS AND MAINTENANCE PROGRAM

This chapter describes the acceptance tests and maintenance program to be used for the NAC-STC to assure compliance with 10 CFR 71 and IAEA Safety Series No. TS-R-1 acceptance and maintenance criteria. Also included is a general description of the fabrication of the NAC-STC cask and the transportable storage canister with additional information on the lead pouring requirements and procedures.

Where required, specific procedures for inspection, special processes, and testing will be developed to support the entire manufacturing process with a Quality Assurance (QA) program that has been approved in accordance with 10 CFR 71 Subpart H and IAEA Safety Series No. TS-R-1.

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8.1 Fabrication Requirements and Acceptance Tests

This section identifies the fabrication, inspection and acceptance requirements, tests, and acceptance criteria established for the NAC-STC to verify, prior to acceptance and packaging marking per 10 CFR 71.85(c), that the packaging has been fabricated, assembled, tested, inspected and accepted in accordance with the applicable NAC-STC License Drawings (Section 1.3.2) and the other requirements of this application.

8.1.1 Weld Procedures, Examination, and Acceptance

The primary containment components of the package and the canister shell will be fabricated in accordance with the ASME Boiler and Pressure Vessel Code (ASME Code), Section III, Division I, Subsection NB requirements. The noncontainment components of the packaging, except the fuel basket assembly and the neutron shield vessel and fins, will be fabricated in accordance with ASME Code, Section VIII, Division 1 requirements. The fuel basket assembly components for either the directly loaded fuel basket or the canister fuel basket will be fabricated in accordance with ASME Code, Section III, Subsection NG. The fabrication of the neutron shield shell and fins, the heat transfer disks and the GTCC basket will be in accordance with ASME Code, Section III, Subsection NF. The fabrication and welding requirements for the NAC-STC cask, and the transportable storage canister, are shown on the NAC-STC License Drawings. Fabrication of the transportable storage canister is described in Section 8.1.8.

In the fabrication of the NAC-STC, the plates and forgings that comprise the cask body are joined by welding. The welding procedure qualifications and the welding performance qualifications for the fabrication of the NAC-STC will be in accordance with Part QW-Welding, Section IX of the ASME Code. All exposed welds on the NAC-STC will be ground flush to the base metal or to a smooth fillet.

Fabricators of the NAC-STC will be experienced and qualified with nuclear component fabrication. Fabrication will be in full accordance with the applicable requirements of ASME Code Section III for the containment vessel, recognizing that design specification, design report, authorized inspection agency, and code stamping do not apply. The fabricator will establish a detailed written weld inspection plan, in accordance with an approved Quality Assurance program, of visual, dye penetrant, ultrasonic and radiographic weld examinations to be performed during fabrication and prior to acceptance of the cask. The weld inspection plan will identify the welds to be examined, the sequence of the examinations, the weld examination

method used, and the criteria for acceptance of the weld in accordance with the applicable sections of the ASME Code.

The finished surfaces of all welds on the NAC-STC will be visually examined in accordance with ASME Code, Section V, Article 9, to verify that the components are assembled in accordance with the License Drawings and that the components are free of nicks, gouges, or other damage. The acceptance criteria for the visually examined welds will be in accordance with ASME Code, Section VIII, Division 1, UW-35 and UW-36.

The NAC-STC primary containment boundary welds shall be radiographic examined in accordance with ASME Code, Section V, Article 2. Acceptance criteria for radiographic examinations shall be in accordance with ASME Code Section III, Division I, Subsection NB, Article NB-5320. Unacceptable imperfections such as a crack, a zone of incomplete fusion or penetration, elongated indications with lengths greater than specified limits, and rounded indications in excess of the limits specified shall be cause for rejection of the weld. Repair of unacceptable weld metal defects in welds shall be in accordance with paragraph NB-4450, of the ASME Code, Section III. Repaired welds shall be reexamined in accordance with the original examination criteria.

The circumferential and longitudinal welds of the outer shell assembly, and the connection welds of the outer shell assembly to the upper forging shall be radiographic examined in accordance with the ASME Code, Section V, Article 2. Acceptance criteria for radiographic examinations of the outer shell welds shall be in accordance with ASME Code, Section VIII, Division 1, UW-51. Repair of unacceptable defects shall be in accordance with UW-38. Repaired welds shall be re-examined in accordance with the original examination criteria.

The final NAC-STC cask closure welds of the bottom ring forging to the bottom forging and to the outer shell, and the closure weld of the bottom plate to the bottom ring forging shall be ultrasonic examined in accordance with ASME Code, Section V, Article 5. Acceptance criteria for the ultrasonic examination shall be in accordance with ASME Code, Section VIII, Division 1, UW-53 and Appendix 12. Repair of unacceptable defects shall be in accordance with UW-38. Repaired welds shall be re-examined in accordance with the original examination criteria.

Welds that are marked "PT Root and Final Pass" on the NAC-STC License Drawings (Section 1.3.2) will be liquid penetrant (PT) examined in accordance with ASME Code, Section V, Article 6. The liquid penetrant examination method is used to detect discontinuities, such as cracks, seams, laps, laminations and porosity, that are open to the surface of nonporous metals. Acceptance criteria for liquid penetrant examined welds shall be in accordance with ASME Code, Section III, Division I, Subsection NB, Article NB-5350. All other noncontainment welds that are marked "PT" on the NAC-STC License Drawings will be liquid penetrant examined in accordance with ASME Code, Section V, Article 6. Acceptance criteria for these noncontainment welds shall be in accordance with ASME Code, Section VIII, Division 1, Appendix 8, except for the fin to outer shell, and fin to neutron shield closure plate welds which will be examined and evaluated in accordance with ASME Code, Section III, Subsection NF, Article NF-5350. Unacceptable indications shall be cause for rejection of the welds. Rejected welds shall be repaired in accordance with approved weld repair procedures prepared in accordance with the applicable provisions of the ASME Code, Section III, NB-4450, for containment welds, NF-4450 for fin to neutron shield shell and cask outer shell welds, and Section VIII, UW-38 for other non-containment welds. Repaired welds shall be re-examined in accordance with the original examination criteria.

All weld inspections shall be performed by qualified personnel in accordance with written procedures. Inspection personnel shall be qualified in accordance with SNT-TC-1A, "Personnel Qualifications and Certification in Nondestructive Testing," the American Society for Nondestructive Testing, Inc., edition as invoked by the applicable ASME Code.

8.1.2 Structural and Pressure Tests

8.1.2.1 Lifting Trunnion Load Testing

Each of the two pairs of the cask lifting trunnions shall be load tested in accordance with the requirements of ANSI N14.6 "Special Lifting Devices for Shipping Containers Weighing 10,000 pounds (4500 kg) or More for Nuclear Materials." The load test will be performed for one pair and repeated for the other pair. The load test shall be performed in accordance with approved written procedures.

The lifting trunnion load test shall consist of applying a vertical load of 750,000 pounds, which is 300 percent of the maximum service load, to diametrically opposite trunnion pairs.

The load will be applied in a vertical direction, equally distributed between the two trunnions and over the length of 2.25 inches of the trunnion/lifting yoke interface areas. The inner and outer lids will be bolted in place for the test. The test may be carried out by the use of calibrated hydraulic rams combined with a load spreading beam, or the cask lifting yoke, attached to the trunnion pair. The load will be held for a minimum of 10 minutes.

Following completion of the lifting trunnion load test, all accessible trunnion welds and load bearing surfaces shall be visually inspected for permanent deformation, galling or cracking. Inspections utilizing liquid penetrant examination shall be performed in accordance with the ASME Code, Section V, Article 6. Liquid penetrant acceptance standards shall be in accordance with Paragraph NF-5350 of the ASME Code, Section III, Division 1.

Any evidence of permanent deformation, cracking, galling of the load bearing surfaces or unacceptable dye penetrant results shall be cause for rejection of the trunnion or related welds.

8.1.2.2 Load Testing of the Rotation Trunnion Recesses

The rotation trunnion recesses at the lower end of the cask shall be load tested. The load test shall be performed in accordance with approved written procedures.

The load test for recesses shall consist of applying a vertical load of 375,000 pounds (170.1 MT) to the rotation trunnion recess pair, which is 150 percent of the maximum service load. The load will be applied in a vertical direction and equally distributed between the two rotation trunnion recesses by the use of hydraulic rams combined with a load-spreading beam.

Following completion of the rotation trunnion recesses load test, all accessible trunnion recess welds and load bearing surfaces shall be visually inspected for permanent deformation, galling or cracking. Inspections utilizing liquid penetrant examination shall be performed in accordance with the ASME Code, Section V, Article 6. Liquid penetrant acceptance standards shall be as indicated in paragraph NF-5350 of the ASME Code, Section III, Division 1.

Any evidence of permanent deformation, cracking, galling of the load bearing surfaces or unacceptable dye penetrant results shall be cause for rejection of the rotation trunnion recesses or related welds.

8.1.2.3 Hydrostatic Testing

A hydrostatic test shall be performed on the NAC-STC cask containment boundary, prior to final acceptance of the cask, in accordance with the ASME Code, Section III, Division I, Article NB-6200. The hydrostatic test pressure shall be at least 76 psig, which is 150 percent of the Maximum Normal Operating Pressure. This test shall be performed in accordance with approved written procedures. All pressure retaining components, appurtenances, and completed systems shall be pressure tested.

The vent port will be used for the test connection. Only the vent port quick-disconnect will be installed during the testing. The hydrostatic test will be performed with the inner lid and the drain port coverplate installed and torqued.

The hydrostatic test system components, although not part of the cask containment boundary, will be visually inspected prior to the start of the hydrostatic test. Leakage from the valves or connections will be corrected prior to the start of the hydrostatic test.

The test pressure gauge installed on the cask will have an upper limit of approximately twice that of the test pressure. The hydrostatic test pressure shall be maintained for a minimum of 30 minutes, during which time a visual inspection is made to detect any evidence of a leak. Any evidence of a leak during the minimum hold period will be cause for rejection.

After completion of the hydrostatic test, the cask containment boundary will be dried and prepared for visual and/or dye penetrant inspections as appropriate. The components of the cask containment boundary shall be visually inspected. All accessible welds within the cavity shall be liquid penetrant inspected. Any evidence of cracking or permanent deformation is cause for rejection of the affected component.

8.1.2.4 Pneumatic Bubble Testing of the Neutron Shield Tank

A pneumatic bubble test of the neutron shield tank will be performed in accordance with Section V, Article 10, Appendix I, of the ASME Code following final closure welding of the bottom closure plates. The pneumatic test pressure shall be $12.5 + 1.5/-0$ psig, which is 125 percent of the relief valve set pressure. The test shall be performed in accordance with approved written procedures.

During the test, the two relief valves on the neutron shield tank will be removed. One of the relief valves threaded connections will be used for connection of the air pressure line and test pressure gauge. The other relief valve connection will be plugged with a threaded plug.

Following introduction of pressurized air into the neutron shield, a 15-minute minimum soak time will be required. Following completion of the soak time, approved soap bubble solution will be applied to all fin to shell, shell to end plate, and end plate to outer shell welds. The acceptance criteria for the bubble test will be no air leak from any tested weld as indicated by continuous bubbling of the solution. If an air leak is indicated, the weld shall be repaired in accordance with approved weld repair procedures and the pneumatic bubble test shall be repeated until no unacceptable air leak is observed.

8.1.3 Leak Tests

Leak tests shall be performed in accordance with Section 7.3 of ANSI N14.5-1997, containment System Fabrication Verification, on the NAC-STC cask containment boundary seals to verify proper fabrication of the cask. The leak tests shall be performed in accordance with approved written procedures. Leak tests shall be performed on the cask containment weldment, the inner lid O-rings, the inner lid interseal test port plug, the vent port coverplate O-rings and its interseal test plug, and the drain port coverplate O-rings and its interseal test plug.

Following completion and NDE of the containment weldment, a helium leak test of the containment weldment shall be performed in accordance with the requirements of ASME Code, Section V, Article 10. The containment weldment shall have an indicated leak rate of less than 2×10^{-7} cm³/sec (helium), using a minimum test sensitivity of 1×10^{-7} cm³/sec (helium). If a leak is detected, the affected weld shall be rejected. Rejected welds shall be repaired in accordance with the requirements of ASME Code, Section III, Division I, Subsection NB,

Article NB-4450. The repaired weld area shall be retested and reinspected in accordance with the above test requirements and acceptance standards.

The containment boundary closures may use either metallic O-rings or nonmetallic Viton O-rings. The two O-ring types require different O-ring groove designs and, therefore, may not be used interchangeably and must be used with the inner lid, vent and drain port coverplates and outer lid having the appropriate O-ring groove machined in the component. The two O-ring types also have different allowable leak rate criteria as described in Section 4.1. Consequently, different acceptance criteria are applied to the metallic and nonmetallic O-ring configurations.

The detailed procedures for the NAC-STC cask leak testing are presented in Section 7.4.

Metallic O-ring Testing

The final fabrication verification leak testing of the containment boundary closures using metallic O-rings consists of a series of leak tests using (minimum 99.9 percent pure) helium as a tracer gas and a helium leak detection system calibrated to a minimum sensitivity of 1×10^{-7} cm³/sec (helium).

The test plug O-rings on the coverplates and the interseal test plug will be tested using the vacuum air pressure rise method. The metallic O-rings in the inner lid and the vent and drain port coverplates will be tested to ensure that the leakage rate is $\leq 2 \times 10^{-7}$ cm³/sec (helium). The tracer gas shall be introduced on the containment side of the O-ring in all cases. The test procedures and methods will be selected to ensure that the sensitivity of each leak test is $\leq 1 \times 10^{-7}$ cm³/sec (helium).

A leak rate past any seal or closure that exceeds 2.0×10^{-7} cm³/sec (helium) shall be cause for rejection of the item being tested. Seal replacement or other corrective actions will be taken to correct the leak. The item shall then be retested and inspected in accordance with the above test requirements and acceptance standards.

Viton O-ring Testing

The final fabrication verification leak testing of the containment boundary closures using Viton O-rings consists of leak tests of the closure O-rings using (minimum 99.9 percent pure) helium as a tracer gas and a helium leak detection system calibrated to a minimum sensitivity of $9.3 \times 10^{-5} \text{ cm}^3/\text{sec}$ (helium).

The test plug O-rings on the coverplate and the interseal test plug will be tested using the vacuum air pressure rise method. The Viton O-rings in the inner lid and the vent and drain port coverplates will be tested to ensure that the leak past all three O-rings will not exceed a cumulative total of $9.3 \times 10^{-5} \text{ cm}^3/\text{sec}$ (helium). The tracer gas shall be introduced on the containment side of the O-ring in all cases. The test procedures and methods will be selected to ensure that the sensitivity of each leak test is $4.7 \times 10^{-5} \text{ cm}^3/\text{sec}$ (helium) or better.

A leak rate past any seal or closure, or the cumulative leakage rate of all containment seals, exceeding $9.3 \times 10^{-5} \text{ cm}^3/\text{sec}$ (helium) shall be cause for rejection of the item(s) being tested. Seal replacement or other corrective actions will be taken to correct the leak. The item shall then be retested and inspected in accordance with the above test requirements and acceptance standards.

8.1.4 Component Tests

Tests performed on individual components are designed to ensure that the component meets the design requirements for correct and proper operation of the cask system.

Acceptance criteria are established based on the functions and design requirements of the component being tested.

8.1.4.1 Valves

There are no valves that are part of the NAC-STC containment boundary for transport. Quick-disconnects are installed in the vent, drain and interseal test port openings in the inner lid to provide access to the cavity, and in the interlid port to provide access to the interlid region. These fittings serve as valves when the mating parts are connected, and are used to connect ancillary equipment to the cask cavity for filling, draining, drying, backfilling, gas sampling, and

leak testing operations. Upon removal of the external fitting, the valve in the quick-disconnect closes automatically. The design and selection of the quick-disconnects is based on similar equipment and procedures used with other NRC-approved storage and transport casks. For transport, the quick-disconnects are sealed inside the transport containment boundary using a bolted coverplate fitted with two O-ring seals.

There are no rupture disks on the NAC-STC.

Two self-actuating pressure relief valves are installed on the external shell of the neutron shield to provide for venting of vapor from the shielding material during transport thermal accident conditions. These valves have stainless steel bodies and an operating pressure range of zero to 200 psig with an adjustable cracking pressure within this range. The cracking pressure is set at 10 psig. These relief valves do not provide a safety function, but have been designed to minimize recovery efforts in the unlikely event of a neutron shield overpressure condition.

8.1.4.2 Gaskets

As described in Section 8.1.3, the containment boundary of the NAC-STC may use either metallic O-rings or non-metallic Viton O-rings. The two O-ring types require different O-ring groove designs and, therefore, may not be used interchangeably and must be used with the inner lid, vent and drain port coverplates and outer lid having the appropriate O-ring groove machined in the component. Metallic O-rings must be used for direct loading of the NAC-STC with fuel for extended storage and for loading of a transportable storage canister (for transport). For direct loading of fuel for immediate transport, either metallic or non-metallic O-rings may be used.

The outer lid, inner lid, drain port coverplate, vent port coverplate, interlid port cover, pressure port cover, and interseal test plug gaskets are O-rings. For transport after an extended period of storage, the containment boundary is formed by the outer metallic O-ring of the inner lid, the outer metallic O-rings on the vent and drain port coverplates, and the interseal test plug metallic O-rings for the inner lid, the vent port coverplate and the drain port coverplate. The inner metallic O-rings of the inner lid, vent port coverplate and drain port coverplate, the metallic O-ring of the outer lid, and the PTFE O-rings of the interlid and pressure port covers provide a secondary closure to the cask contents. For immediate transport, the containment boundary is formed by the inner O-rings of the inner lid and vent and drain port coverplates. A second boundary is formed by the O-rings of the outer lid and interseal and pressure port covers.

The O-ring replacement schedule depends upon the O-ring material. The metallic O-ring(s) of any component shall be replaced prior to reinstallation of the component. Viton O-rings are inspected prior to each use and replaced as necessary. The PTFE O-rings of the interlid and pressure ports will be visually inspected prior to each use, and replaced if necessary. The PTFE O-rings shall be replaced at least once every two years during cask transport operations, or prior to transport if they have been installed longer than two years (i.e., after extended storage).

The containment boundary O-ring shall be tested and maintained in accordance with the Maintenance Program Schedule of Table 8.2-1 and the leak test criteria of Section 8.2.2.

8.1.4.3 Miscellaneous

The removable transport impact limiters consist of redwood and balsa wood. License drawings and the supporting analyses specify the crush strengths of the redwood and balsa wood to be $6240 \text{ psi} \pm 620 \text{ psi}$ and $1550 \text{ psi} \pm 150 \text{ psi}$ respectively. For manufacturing purposes, verification of the impact limiter material is accomplished by verifying the densities of the wood. Three samples from each redwood board are to be tested for density, and the average density of the samples shall be 23.5 ± 3.5 pounds/cubic foot. Each 15-degree and 30-degree pie shaped section of the impact limiter shall have a density of 22.3 ± 1.2 pounds/cubic foot in accordance with the License Drawings. The moisture content for any single redwood board must be greater than 5 percent, but less than 15 percent. The average moisture content for a lot of redwood used in impact limiter construction must not be greater than 12 percent.

Following final closure welding of the transport impact limiter stainless steel shell, a leak test of the shell welds shall be performed to verify weld integrity. The test shall be performed by evacuating the impact limiter to 75 mbar and performing a 30-minute test to determine if there is any increase in the impact limiter pressure. Any detected leak shall not exceed $1 \times 10^{-2} \text{ cm}^3/\text{sec}$. If a leak exceeding this value is detected, the cause of the leak shall be determined, and the weld repaired and retested.

8.1.5 Tests for Shielding Integrity

8.1.5.1 Gamma Shield Test

The gamma scan test shall be conducted by continuous scanning or probing over 100 percent of all accessible cask surfaces using a 3-inch detector and a ^{60}Co source. The source strength shall be of an intensity sufficient to produce a count rate that equals or exceeds three times the background count rate on the external surfaces of the cask. The count rate shall be maintained for greater than one minute prior to the start of scanning. The detector scan path spacing (cask exterior surface) will be a maximum of 2.5 inches and the scanning speed will be 4.5 feet per minute or less. The source scan path spacing (cask interior surface) will be on a 2-inch grid pattern (when using a 3-inch detector). Flat surfaces, such as the cask bottom and closure lids, will use a 2.5 inch spacing for both the detector and source scan paths (when using a 3-inch detector).

The acceptance criteria for the shield test will be that the shield effectiveness of the cask body and lids shall be equal to or greater than the shield effectiveness of a lead and steel mock-up. The steel thickness of the mockup shall be equivalent to the minimum steel thickness specified on the License Drawings and the lead thickness shall be equivalent to the minimum lead thickness specified in the License Drawings less 3 percent. The shielding mock-up will be produced using the same fabrication techniques as those approved for the cask.

Measured count rates that exceed those established by the test mock-up shall cause the component to be rejected. The rejected areas/components shall be evaluated to determine the corrective action to be taken. Any repaired areas shall be retested prior to acceptance.

An additional gamma shield effectiveness test shall be performed on each cask following first fuel loading. The neutron and gamma shield effectiveness test procedures and acceptance criteria are described in Section 8.1.5.4.

8.1.5.2 Neutron Shielding Test

The neutron shielding of the NAC-STC is provided by a solid layer of NS-4-FR, which is a hard polymer material. A 5.5-inch layer of NS-4-FR is located in the annulus formed by the outer shell and the 0.236-inch (6 mm) thick neutron shield shell. The neutron shield is divided in sections by the copper/stainless steel fins. A 2-inch thick layer of NS-4-FR is also installed in the cask inner lid and in the cask bottom.

The installation of NS-4-FR material in the fabrication of the cask is a special process and, as such, procedures will be prepared and qualified to ensure that the mix ratios, mixing method, degassing, pouring, and curing of the material is properly performed. The NS-4-FR raw material is provided in the form of a 3-part mixing kit. The material content of the raw material is tested and certified at the time of kit preparation. The neutron shielding material is installed into the annulus between the outer shell and the neutron shield shell by pouring it with the cask in an inverted vertical position. Prior to installation, samples from each mix of the actual material being poured into the annulus are wet density tested to ensure that the material is properly mixed. Mixes that do not meet the wet density acceptance criteria are rejected. Procedures used for installation of the material are validated prior to use by destructive examination of a full scale mock-up of the neutron shield cavity. Qualification of the installation procedure verifies material homogeneous properties and minimizes the potential deleterious voids.

8.1.5.3 Neutron Shielding Material Testing

The neutron shield properties of NS-4-FR are provided in Chapters 1 and 3. Each lot (mixed batch) of neutron shield material shall be tested to verify that the material composition (aluminum and hydrogen), boron concentration, and neutron shield density meet the requirements specified in Chapters 1 and 3 and the License Drawings. Testing shall be performed by qualified laboratories in accordance with written and approved procedures. Material composition, boron concentration, and density data for each lot of neutron shield material shall become part of the quality record documentation package.

Dimensional inspection of the cavities containing the neutron shielding material shall ensure that the required thickness specified in the License Drawings is incorporated into the cask.

The installation of the neutron shielding material shall be performed in accordance with written, approved, and qualified procedures. The procedures shall ensure that mix ratios and mixing methods are controlled in order to achieve proper material composition, boron concentration and distribution, and that pours are controlled in order to prevent gaps or unacceptable voids from occurring in the material. Procedures shall be qualified by the use of mock-ups to ensure that the NS-4-FR installation does not result in the creation of unacceptable voids. Samples of each lot of neutron shield material shall be maintained as part of the quality record documentation package.

8.1.5.4 Neutron and Gamma Shield Effectiveness Tests

Following first fuel loading, a neutron and gamma shield effectiveness test shall be performed for each cask prior to transport. The test shall be performed with the cask loaded with fuel, drained, vacuum dried and backfilled with helium. The purpose of the test is to document the effectiveness of the neutron and gamma shielding materials. The test shall be performed in accordance with detailed, approved written test procedures.

Calibrated neutron and gamma dose rate meters shall be used to measure the neutron and gamma dose rate at contact with the outer shell of the neutron shield and at 2.3 meters from the surface (equivalent to 2 meters from the sides of the railcar). Dose measurement points shall be established on the external surface of the shell at 30° intervals and at five points along the height of the shield (a total of 60 measuring points). In addition, neutron and gamma dose rate measurements shall be made of the trunnion areas above the neutron shield, at four points below the neutron shield, and at the edges and center of the cask top (outer lid) and cask bottom surfaces. Dose rates at the top and bottom of the cask shall be measured with the transport impact limiters installed. The dose rates measured at contact and at 2.3 meters shall be recorded on the test data sheet, along with the total power of the loaded fuel assemblies; date, time and location of test; identification and calibration of instrumentation; and identification of test engineer and operators.

To allow an evaluation of the measured dose rates to be completed, the burnup and cool time for the actual fuel assemblies loaded into the cask will be determined and recorded. From this fuel history data, the total actual neutron and gamma source terms will be estimated using ORIGEN or similar calculations.

If the measured dose rates exceed the applicable regulatory limits, the licensee shall notify the NRC. Appropriate corrective measures will be taken, including fuel unloading and correction of the shielding deficiency. Following corrective actions, the test will be reperformed to the original acceptance criteria prior to final acceptance.

8.1.6 Thermal Test

Prior to acceptance at the factory, a thermal test shall be performed on each fabricated packaging to confirm and verify that the fabricated and assembled cask possesses the heat rejection capabilities predicted by the thermal analyses. The thermal test shall be performed in accordance with approved written procedures.

8.1.6.1 Thermal Test Set-up

The thermal test set-up is shown in Figure 8.1-1(a). As depicted, the thermal test shall be performed with the cask positioned horizontally on a test frame. The transport impact limiter or equivalent insulating material shall be installed on each end of the cask to simulate the transport configuration. The cask will be located in a covered building in a still environment. The cask shall be assembled with the basket installed. A thermal test lid with connections for thermocouple leads and electric heater power cables shall be installed in place of the inner lid. The outer lid will not be installed for the test. The thermal test lid will be provided with an O-ring seal capable of containing the containment cavity helium atmosphere.

Electric heaters shall be installed in each fuel tube. The electric heaters will have an active length of between 120 and 150 inches and be capable of generating a minimum of 22 kilowatts (kw). The heaters will be supported in the basket so as to not be in contact with the wall of the fuel tube. The power supplied to the heater will be recorded throughout the test duration.

Calibrated test thermocouples, with an accuracy of $\pm 2^{\circ}\text{F}$, will be installed on the cask basket, inner shell, and outer neutron shield shell surfaces. The location of the test thermocouples are shown in Figure 8.1-1. The specific location of the thermocouples are as follows:

- TC1 - basket top steel weldment
- TC2 - steel disk at cask basket midpoint
- TC3 - aluminum disk at cask basket midpoint
- TC4 - basket bottom steel weldment
- TC5; TC6; TC7; and TC8 - located at 90° intervals on the inner shell surface at cavity midpoint
- TC9 - top of inner shell surface at 30-40 inches from top of cavity
- TC10 - bottom of inner shell surface at 30 to 40 inches from base of the cavity
- TC11; TC12; TC13; and TC14 - located at 90° intervals on the neutron shield shell surface (at fin tip) at cask midpoint.
- TC15 - top of neutron shield shell surface (at fin tip) at 30-40 inches from top of neutron shell.
- TC16 - bottom of neutron shield shell surface (at fin tip) at 30-40 inches from bottom of neutron shield shell.
- TC17 - top of upper forging

- TC18 - outer shell surface at centerline of cask bottom face
- TC19 - inner fuel tube wall surface near the center of the cask basket
- TC20 - ambient temperature of testing area

The output of the test thermocouples will be recorded throughout the test by a strip chart recorder.

8.1.6.2 Test Procedure

With the cask assembled and instrumented as described above, the cask cavity is evacuated and backfilled to 1.0 atmosphere absolute (14.6 psia) with helium. Power will be applied to the heaters to simulate the cask contents. After initiation of power to the heaters, the temperatures of all thermocouples and heater power levels will be monitored and recorded on data sheets at 60 minute intervals. Power will be maintained to the electrical heaters until the cask has reached thermal equilibrium.

For the purpose of the test, thermal equilibrium is defined as being achieved when over two consecutive hours:

$$\Delta t_{TC13} \leq 2^{\circ}\text{F/hr, and}$$
$$\Delta t_{TC3} \leq 2^{\circ}\text{F/hr}$$

Based upon the thermal heat-up evaluation, thermal equilibrium should be achieved in approximately five days.

After verification of thermal equilibrium, final temperature measurements will be recorded for all test thermocouples. The final power readings for the electric heaters will also be recorded. The strip chart will be marked to indicate the time of the final cask measurements. The printout of the strip chart recorder and the completed test data sheets will be incorporated into an approved final thermal test report. The test will be determined to be acceptable if the acceptance criteria of Section 8.1.6.3 are met.

If the acceptance criteria are not met, the cask will not be accepted until appropriate corrective actions are completed. Upon completion of corrective actions, the cask shall be retested to the original test requirements and acceptance criteria.

8.1.6.3 Acceptance Criteria

The purpose of the thermal test is to confirm the heat rejection capabilities of the as-built cask are acceptable and correspond to the temperatures calculated by thermal analyses for the directly loaded (uncanistered) configuration presented in Chapter 3.0 of this application.

Package heat dissipation acceptance testing assures: 1) maximum material temperatures do not exceed material allowables; and that 2) measured temperature gradients are less than the thermal gradients calculated in the package thermal analyses.

The thermal acceptance test is accepted when the following criteria are met:

- 1) When corrected for physical test boundary conditions and heat load, the following measured temperatures are not exceeded:

<u>TC No.</u>	<u>Location</u>	<u>Temperature °F</u>
TC1	Top Basket Steel Weldment	435
TC3	Aluminum Disk Center	485
TC2	Steel Support Disk Center	495
TC4	Basket Bottom Steel Weldment	475
TC5-TC8	Cask Inner Shell	330
TC11-TC14	Neutron Shield Shell	240
TC17	Cask Top Forging	200
TC18	Cask Bottom	330
TC19	Tube Wall	540

- 2) The measured temperature gradient across the central steel disk from TC2 to the average of TC5, TC6; TC7 and TC8 is less than 200°F;
- 3) The measured temperature gradient across the central aluminum disk from TC3 to the average of TC5; TC6; TC7 and TC8 is less than 190°F; and
- 4) The measured temperature gradient across the cask body as measured by thermocouple pairs TC5-TC13; TC6-TC14; TC7-TC11; and TC8-TC12 are less than 90°F.

8.1.7 Neutron Absorber Tests

Two alternate neutron poison materials, BORAL and TalBor, have been qualified by NAC for use in the fuel tubes. BORAL is manufactured by AAR Advanced Structures (AAR), under a

Quality Assurance/Quality Control program in conformance with the requirements of 10 CFR 50, Appendix B. The manufacturing process consists of several steps: the first step is the mixing of the aluminum and boron-carbide powders that form the core of the finished material, with the amount of each powder a function of the desired ^{10}B areal density. The methods used to control the weight and blend of the powders are patented and proprietary processes of AAR. The mixture of powders is placed in an aluminum box with walls approximately one inch thick. The top lid is welded in place. This "ingot" is heated for several hours and then is hot-rolled to produce the sheet of design thickness. The rolling process densifies and bonds the powder mixture. The aluminum box walls become the cladding for the Al-B₄C core.

TalBor is manufactured by Talon Composites, Inc. (TalBor was formerly called Boralyn, and was produced by Alyn Corporation. Alyn Corporation went out of business and Talon Composites acquired the major production equipment and the patent rights for Boralyn. TalBor is essentially identical to Boralyn.) TalBor is manufactured and controlled using a Quality Assurance program that is compliant with the applicable requirements of 10 CFR 50, Appendix, B. TalBor is a metal matrix composite (MMC). The aluminum and B₄C powders are mixed to the specified ^{10}B areal density and the powder mixture is vacuum sintered and hot pressed to achieve a fully dense billet. The billet is extruded, then cut and rolled to the design thickness.

After manufacturing, test samples from each batch of neutron absorber (poison) sheets shall be tested to verify the presence, proper distribution, and minimum weight percent of ^{10}B . Neutron transmission testing or augmented wet chemistry testing may be used. The tests shall be performed in accordance with approved written procedures.

8.1.7.1 Neutron Absorber Material Sampling Plan

The neutron absorber sampling plan is selected to demonstrate a 95/95 (95% probability and 95% confidence level) statistical confidence level in the neutron absorber sheet material compliance with the specification. In addition to the specified sampling plan, each sheet of material is visually and dimensionally inspected using at least 6 measurements (along the edges near each corner and the longitudinal centerline) on each sheet. No rejected neutron absorber sheet is used. The sampling plan is supported by written and approved procedures.

The sampling plan requires that a coupon sample be taken from each sheet of the first set of 100 sheets of absorber material. Thereafter, coupon samples are taken from 20 randomly selected sheets from each set of 100 sheets. This 1 in 5 sampling plan continues until there is a change in lot or batch of constituent materials of the sheet (i.e., boron carbide powder, aluminum powder,

or aluminum extrusion), or a process change, at which time the sampling process is reinitiated as previously described. The sheet samples are indelibly marked and recorded for identification. This identification is used to document neutron absorber test results, which become part of the quality record documentation package.

8.1.7.2 Wet Chemistry Test Performance

An approved facility with chemical analysis capability shall be selected to perform the wet chemistry tests. The tests will ensure the presence of boron and enable the calculation of the ^{10}B areal density. Acceptability of the uniformity of boron distribution is based on the manufacturer's material qualification tests.

The most common method of verifying the acceptability of neutron absorber material is the wet chemistry method—a chemical analysis where the aluminum is separated from a sample with known thickness and volume. The remaining boron-carbide material is weighed and the areal density of ^{10}B is computed. A statistical conclusion about the BORAL or TalBor sheet from which the sample was taken and that batch of sheets may then be drawn based on the test results and the established manufacturing processes previously noted.

8.1.7.3 Neutron Transmission Test Performance

An approved facility with a neutron source and neutron detection capability shall be selected to perform the described tests, if the neutron transmission test method is used. The tests will assure that the neutron absorption capacity of the material tested is equal to, or higher than, the given reference value and will verify the uniformity of boron distribution. The principle of measurement of neutron absorption is that the presence of boron results in a reduction of neutron flux between the thermalized neutron source and the neutron detector—depending on the material thickness and boron content.

Typical test equipment will consist of thermal neutron source equipment, a neutron detector and a counting instrument. The test equipment is calibrated using a known standard, whose ^{10}B content has been checked and verified by an independent method such as chemical analysis. This calibration process shall be repeated daily (every 24 hours) while tests are being performed.

8.1.7.4 Acceptance Criteria

The neutron transmission test results shall be considered acceptable if the minimum ^{10}B areal density is determined to be equal to, or greater than, that specified on the fuel tube drawings.

Any specimen not meeting the acceptance criteria shall be rejected and all of the sheets from that batch shall be similarly rejected unless coupons from each individual absorber plate are tested and confirmed to meet or exceed the specified areal density.

8.1.8 Transportable Storage Canister

The transportable storage canister is constructed of Type 304L stainless steel and is fabricated by welding. If circumferential welds are required to join two shell sections, the seam welds shall not be aligned within 45° circumferentially. The welded cylinder is closed at the bottom by a circular plate welded to the shell wall. The top of the cylinder is closed by two field-installed circular plates, welded to the canister shell wall following fuel loading.

The transportable storage canister is a welded closed component. The canister serves as the confinement boundary component of the NAC-MPC System during storage of spent fuel in the vertical concrete cask.

The finished surfaces of all canister welds are visually examined in accordance with ASME Code Section V, Article 9, to verify that the components are assembled in accordance with the License Drawings and that the components are free of nicks, gouges, and other damage. The acceptance criteria for the visually examined welds is in accordance with ASME Code Section VIII, Division 1, UW-35 and UW-36 and Section III, Subsection NB, NB-4424 and NB-4427.

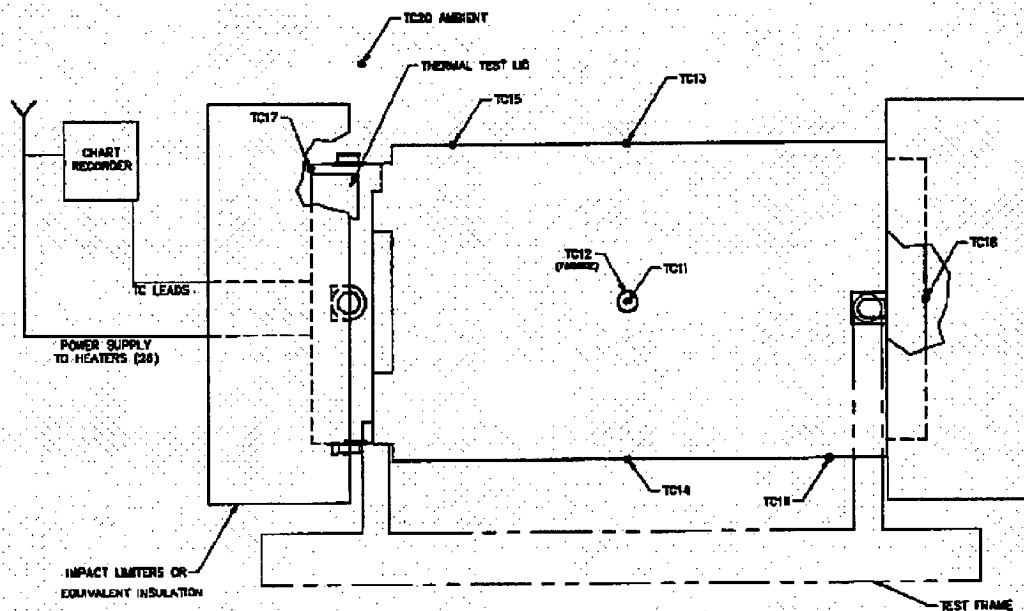
The seam and girth welds in the transportable storage canister shell are full-penetration welds that are radiographic (RT) examined in accordance with ASME Code Section V, Article 2. The acceptance criteria for the RT-examined welds is that specified in ASME Code Section III, Subsection NB, Article NB-5320. The canister shell to bottom plate weld is a full-penetration double-bevel weld with an inside fillet weld that is ultrasonic examined in accordance with ASME Code Section V, Article 5, with acceptance criteria as specified in ASME Code Section III, Subsection NB, Article NB-5330. The final surfaces of the seam and girth welds in the canister and the canister shell to bottom plate weld are also liquid penetrant examined in accordance with ASME Code Section V, Article 6, with the acceptance criteria being that specified in ASME Code Section III, Subsection NB, Article NB-5350.

Field installed partial-penetration groove welds attach the shield (inner) lid to the canister shell and the vent port and the drain port coverplates to the shield lid, after the canister is loaded. The structural lid is attached to the canister shell by a partial penetration weld. The root and final surfaces of the shield lid weld are liquid penetrant examined in accordance with the ASME Code

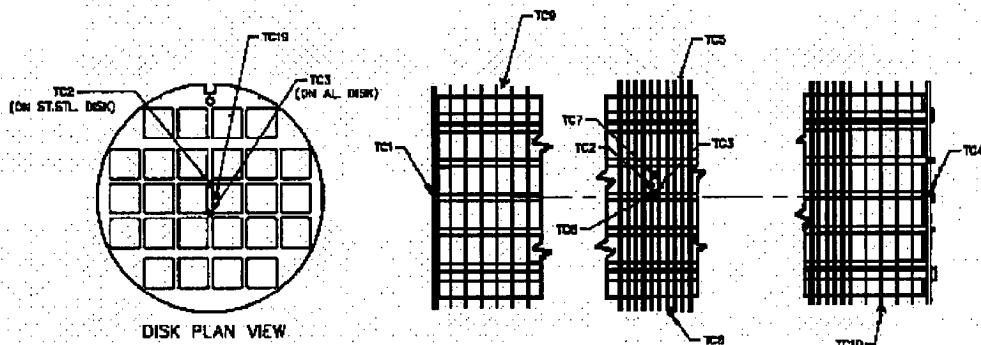
Section V, Article 6. Acceptance criteria are as specified in ASME Code Section III, Division 1, Subsection NB, Article NB-5330. The vent port and the drain port coverplate to shield lid welds are liquid penetrant examined, i.e., root and final surfaces, in accordance with ASME Code Section V, Article 6. Acceptance criteria is specified in ASME Code Section III, Division 1, Subsection NB, Article NB-5350. The structural lid weld is either ultrasonic (UT) examined in accordance with the ASME Code, Section V, Article 5 with the final weld surface liquid penetrant examined in accordance with ASME Code, Section V, Article 6, or progressively liquid penetrant examined in accordance with the ASME Code, Section V, Article 6. Acceptance criteria is specified in ASME Code Section III, Division 1, Subsection NB, Article NB-5330 (ultrasonic) and Article NB-5350 (liquid penetrant). Following completion of the shield lid to canister shell weld and the drain port coverplate to shield lid welds, the canister is leak tested in accordance with ASME Code Section V, Article 10, Appendix V, using a minimum leak rate test sensitivity of $1 \times 10^{-7} \text{ cm}^3/\text{sec}$ (helium).

The fabricator of the transportable storage canister will establish a written weld inspection plan in accordance with an approved quality assurance program. The weld inspection plan will include visual, liquid penetrant, ultrasonic, and radiographic examination. In addition, the weld inspection plan will identify the welds to be examined, the sequence of the examinations, the type of examination method to be used, and the criteria for acceptance of the weld in accordance with the applicable sections of the ASME Code.

Figure 8.1-1 Thermal Test Arrangement



TEST SET-UP AND
(a) EXTERNAL THERMOCOUPLE LOCATIONS



(b) INTERNAL CAVITY AND BASKET THERMOCOUPLE LOCATIONS

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8.2 Maintenance Program

To ensure that the NAC-STC packaging is in compliance with the requirements of the regulations, the Certificate of Compliance, and this application, a cask Maintenance Program for the NAC-STC shall be established. The cask Maintenance Program shall specify the inspections, tests, and replacement of components to be performed, and the frequency and schedule for these activities. This chapter describes the overall requirements of the Maintenance Program and establishes the frequency and schedule for the maintenance activities. The detailed, written inspection, test, component replacement, and repair procedures shall be included in the NAC-STC Operations Manual. The NAC-STC Operations Manual will be issued to Users of the packaging and will be prepared and issued prior to first use of the cask in each configuration.

There are no maintenance requirements for the welded canister containing either fuel or GTCC waste.

8.2.1 Structural and Pressure Tests of the Cask

The four lifting trunnions and the two rotation trunnion recesses shall be visually inspected prior to each shipment. The visual inspections shall be performed in accordance with approved written procedures, and inspection results shall be evaluated against established acceptance criteria.

Evidence of cracking on the load bearing surfaces shall be cause for rejection of the affected trunnion until an approved repair has been completed, and the surfaces re-inspected and accepted. Such repairs shall be implemented and documented in accordance with an approved QA program.

The lifting trunnions are also inspected annually in accordance with Paragraph 6.3.1(b) of ANSI N14.6. All accessible trunnion welds and accessible welds that are part of the load path are visually inspected for permanent deformation, galling, or cracking. Liquid penetrant examinations of welds and load bearing surfaces are performed in accordance with the ASME Code, Section V, Article 6 [3]. Liquid penetrant acceptance standards are those of Paragraph NF-5350 of the ASME Code, Section III, Division 1.

During periods of nonuse of the transport cask, the inspection of the trunnions may be omitted provided that the trunnions are inspected in accordance with this section prior to the next use.

8.2.2 Leak Tests

Leak tests are performed in accordance with the methodologies and requirements of ANSI N14.5-1997, using approved written procedures.

8.2.2.1 Containment Fabrication Verification Leak Test

The containment fabrication verification leak test is performed on each NAC-STC cask at the fabricator's facility in accordance with Section 8.1.3.

8.2.2.2 Containment Periodic Verification Leak Test

The periodic verification leak test shall be performed on each cask after the third use (prior to fourth cask loading sequence) and every twelve months thereafter to verify the containment capability and whenever a replaceable containment component is installed. Metallic O-rings used for the containment boundary seals shall be replaced during each cask loading operation and the seals leak tested in accordance with the containment system periodic verification leak test requirements. Viton O-rings shall be inspected prior to each use and replaced as necessary. Viton O-ring performance shall be demonstrated by leak testing prior to each shipment.

The periodic verification leak test shall be performed using approved written test procedures and in accordance with the test requirements and acceptance criteria established in Section 8.1.3 for the containment fabrication verification leak test.

During periods when the cask is not in use for transport, the periodic verification leak test need not be performed on an annual basis, but shall be reperformed prior to returning the cask to service and use as a transport package.

8.2.2.3 Acceptance Criteria

For the containment verification leak tests, the leak rate for containment boundary metallic O-rings shall be less than or equal to 2×10^{-7} cm³/sec (helium). The minimum test sensitivity for both the fabrication verification and verification leak tests shall be 1.0×10^{-7} cm³/sec (helium).

For Viton O-rings, the maximum (total) permissible leak rate for all containment boundary O-rings shall be less than or equal to 9.3×10^{-5} cm³/sec (helium). The minimum test sensitivity for both the fabrication verification and verification leak tests shall be 4.7×10^{-5} cm³/sec (helium). Unacceptable leak test results shall be cause for rejection of the component tested. Corrective actions, including repair or replacement of the O-rings and/or closure component, shall be taken and documented as appropriate. The leak test shall be repeated and accepted prior to returning the cask to service.

8.2.3 Subsystems Maintenance

There are no subsystems maintenance requirements on the NAC-STC.

8.2.4 Valves, Rupture Disks and Gaskets on the Containment Vessel

There are no valves on the NAC-STC packaging providing a containment function. Four quick-disconnects, one each on the vent, drain, inner lid interseal test and interlid ports, are provided for ease of cask operation.

The quick-disconnect shall be inspected during each cask loading and unloading operation for proper performance and function. As necessary, the subject quick-disconnect shall be replaced. The quick-disconnects shall be replaced every two years during transport operations, and following fuel unloading after extended storage.

There are no rupture disks on the NAC-STC containment vessel.

All O-rings on the NAC-STC shall be visually inspected for damage during each cask operation. All metallic O-rings shall be replaced during each cask loading sequence. PTFE O-rings shall be replaced if damage is noted during the visual inspection and every two years during transport operations. Viton O-rings shall be replaced annually and as required, based on leak testing results and inspections during operations.

8.2.5 Shielding

The gamma and neutron shields of the NAC-STC packaging do not degrade with time or usage. The radiation surveys performed by licensees prior to transport and upon receipt of the loaded cask provide a continuing validation of the shield effectiveness of the NAC-STC.

8.2.6 Periodic Thermal Test

A periodic thermal test program will be established for each operational NAC-STC packaging. During use of the packaging for transport operations, the periodic thermal test will be performed every five years, or prior to the next use if the period exceeds five years. For NAC-STC packagings utilized for extended storage operations exceeding five years, the periodic thermal test will be performed prior to transport. The periodic thermal test shall be performed in accordance with written, approved procedures.

8.2.6.1 Periodic Thermal Test Set-Up

For periodic thermal test performance, the cask will be in a vertical orientation, loaded with spent fuel, and at thermal equilibrium. For the periodic thermal test, thermal equilibrium is defined as a temperature change of $\leq 3^{\circ}\text{F/hr}$ at a single centerline fin tip location.

The decay heat load and fuel cycle history of the fuel assemblies loaded in the cask will be known and recorded on the test data sheets. Thermocouples and/or a surface pyrometer calibrated to an accuracy of $\pm 2^{\circ}\text{F}$ will be used for temperature measurements during the test.

8.2.6.2 Periodic Thermal Test Procedure

With the cask in a vertical orientation located in a cask preparation area or on a storage pad, a temperature measurement will be taken at a marked cask neutron shield shell centerline fin tip and recorded. Repeat temperature measurement at the marked fin tip location until thermal equilibrium criteria of a temperature change of $\leq 3^{\circ}\text{F/hr}$ is met. Upon verification of thermal equilibrium, the test temperature measurements will be performed as follows:

- eight (8) fin tip locations will be marked and identified at 45° intervals in the top one-third of the neutron shield shell;
- eight (8) fin tip locations will be marked and identified at 45° intervals in the central one-third of the neutron shield shell;
- eight (8) fin tip locations will be marked and identified at 45° intervals in the bottom one-third of the neutron shield shell;

- four (4) upper forging surface locations will be marked and identified at 90° intervals above the neutron shield; and
- four (4) bottom forging surface locations will be marked and identified at 90° intervals below the neutron shield.

Temperature measurements will be taken and recorded at the twenty-four (24) marked fin tip locations and recorded on the test data sheets. Temperature measurements of the top forging and bottom forging will also be taken and recorded.

The test results will be reviewed and evaluated to verify the acceptance criteria of 8.2.6.3 are met. The results of the test will be documented in an approved test report and maintained in the packaging's maintenance program records. If the acceptance criteria are not met, the cask will be tagged as non-conforming until corrective actions are taken. Upon completion of the corrective actions, the cask shall be retested to the original periodic thermal test requirements and acceptance criteria.

8.2.6.3 Periodic Thermal Test Acceptance Criteria

The relationship between the temperature of both ends of the cask relative to the average mid-plane temperature approaches unity as the cask heat load decreases from the design bases. The results of the periodic thermal test will be accepted if the test criteria of a), b), c), and d) below are met:

- a) The temperature ratio for the outside surface of the top forging with respect to the average mid-plane neutron shield shell surface temperature meets the test criteria:

<u>Design</u>	<u>Test</u>
$\frac{T_{\text{Top}}}{T_{\text{mid-plane}}} = \frac{170}{243} = 0.7$	$1.0 \geq \frac{T_{\text{Top}}}{T_{\text{mid-plane}}} \geq 0.7$

- b) The ratio of temperatures of the outside surface of the bottom forging with respect to average mid-plane neutron shield shell surface temperature meets the test criteria:

<u>Design</u>	<u>Test</u>
---------------	-------------

$$\frac{T_{\text{Bottom}}}{T_{\text{mid-plane}}} = \frac{280}{243} = 1.2$$

$$1.0 \leq \frac{T_{\text{Bottom}}}{T_{\text{mid-plane}}} \leq 1.2$$

- c) Measured temperatures at the top, bottom, and package mid-plane will be equal to or less than design values when corrected for heat load and ambient temperature in accordance with the following relationship:

$$T_{\text{Actual}} \leq \frac{Q_{\text{Decay}} (T_{\text{Design}} - 100)}{Q_{\text{Design}}} + T_{\text{Test Ambient}}$$

- d) The individual variations of fin tip temperatures around each zone (upper, middle, and lower) do not exceed 20°F from the zone average.

8.2.7 Miscellaneous

The transport impact limiters shall be visually inspected prior to each shipment. The limiters shall be visually inspected for gross damage or cracking to the stainless steel shells in accordance with approved written procedures and established acceptance criteria. Impact limiters not meeting the established acceptance criteria shall be rejected until repairs are performed and the component reinspected and accepted.

The cask cavity shall be visually inspected prior to each fuel loading. Evidence of gross scoring of the cavity surface, or build-up of other foreign matter in the cask cavity that could block the cavity drainage paths shall be cause for rejection of the cask for use until approved maintenance and/or repair activities have been acceptably completed. The basket assembly for the directly loaded (uncanistered) or canistered configuration shall be visually inspected for deformation of the basket disks or tubes. Evidence of damage shall be cause for rejection of the basket until approved repair activities have been completed, and the basket has been re-inspected and approved for use.

The overall condition of the cask, including the fit and function of all removable components, shall be visually inspected and documented during each cask use. Components or cask conditions which are not in compliance with the Certificate of Compliance shall cause the cask to be rejected for transport use until repairs and/or replacement of the cask or component are performed, and the component reinspected and accepted.

The results of the visual inspections, leak tests, shielding and radiological contamination surveys; fuel identification information for the package contents; date, time, and location of the cask loading operations; and remarks regarding replaced components shall be included in the cask loading report for each loaded cask transport. The requirements of the cask loading report shall be detailed in the NAC-STC Operations Manual.

8.2.8 Maintenance Program Schedule

Table 8.2-1 presents the overall maintenance program schedule for the NAC-STC.

Table 8.2-1 Maintenance Program Schedule

Task	Frequency
Cavity Visual Inspection	Prior to Fuel Loading
Basket Visual Inspection	Prior to Fuel Loading
O-ring Visual Inspection	Prior to Fuel Loading
Outer Lid, Inner Lid and Port Coverplate Bolt Visual Inspection	Prior to installation during each use
Cask Visual and Proper Function Inspections	Prior to each Shipment
Lifting and Rotation Trunnion Visual Inspection	Prior to each Shipment
Liquid Penetrant Inspection of surfaces and accessible welds	Annually during use
Maintenance Periodic Leak Rate Test of Inner Lid and Port Coverplate O-rings	For Viton O-rings, annually or when replaced. For metallic O-rings, prior to each shipment
Preshipment Leak Rate Test	Prior to shipment for casks with Viton O-rings
Transport Impact Limiter Visual Inspection	Prior to each shipment
Quick-disconnect Inspection for Proper Function	During each Cask Loading/Unloading Operation
Quick-disconnect Replacement	Every two years during transport operations
Metallic O-ring Replacement	Prior to installation for a loaded transport
Viton O-ring Replacement	Annually, or more often, based on inspection or leak test results
Inner and Outer Lid Bolt Replacement	Every 240 bolting cycles (Every 20 years at 12 cycles per year)
PTFE O-ring Replacement	Every two years during transport operations or as required by inspection
Periodic Leakage Rate Test	Performed within 12 months prior to each shipment for Viton O-rings. No testing needed for out-of-service packaging.
Periodic Thermal Test	Every five years during transport operations, or prior to transport following extended storage periods exceeding five years.

8.3 Quick-Disconnect Valves

“Snap-Tite” quick-disconnect nipples (quick-disconnects) are used in the vent, drain, and inner lid interseal test ports to isolate the cavity, and in the interlid port to isolate the region between the inner and outer lids. No credit is taken for any containment function provided by these components. The drain line quick-disconnect of the Transportable Storage Canister need not be valved.

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8.4 Cask Body Fabrication

8.4.1 General Fabrication Procedures

The NAC-STC cask body is a welded structure of stainless steel plates and forgings. Chemical Copper lead is poured in place between the inner and outer shells to serve as the main gamma shielding material. NS-4-FR is poured in place between the neutron shield shell and the outer shell. NS-4-FR is also form fit between the bottom inner forging and the bottom plate and in the inner lid. Welding on the NAC-STC shall be performed in accordance with the requirements of the ASME Code and the American Welding Society (AWS) Structural Welding Code - Steel (ANSI/AWS D.1-1) as specified on the NAC-STC License Drawings and Section 8.1.1.

The general fabrication procedures for the NAC-STC are summarized, herein, to facilitate an understanding of the component configurations and the weld locations shown on the license drawings.

Each of the two inner shell rings (upper and lower) is rolled from Type XM-19 stainless steel plate and seam welded longitudinally. The outside diameter of each inner shell ring is machined to the defined transition section dimensions. The minimum length of each Type XM-19 shell ring shall be in accordance with the License Drawings. The central inner shell sections are each rolled from Type 304 stainless steel plate and seam welded longitudinally. The number and length of the individual inner shell sections to be used to obtain the required total inner shell length is optional. The inner shell sections are girth welded to each other and the inner shell rings are girth welded on each end of the inner shell. Longitudinal seam welds in adjacent inner shell sections shall be offset 180 degrees for girth-welded sections.

After initial rough machining and final weld preparation, the top forging and the bottom inner forging are individually welded to the opposite ends of the inner shell/inner shell ring weldment to form the cask cavity. The preparation, examination, and acceptance procedures for the welds are described in Section 8.1.1 and defined on the License Drawings. Following inspection and acceptance of the welds, the top forging and the outside diameter of the cask cavity weldment are final machined. Following final machining of both sides of the inner shell, an ultrasonic thickness test of the inner shell wall of the cask cavity shall be performed to confirm that the wall thickness of any location on the shell is not less than 1.46 inches (37.1 mm). A wall thickness at any location of less than 1.46 inches (37.1 mm) will be cause for rejection. Rejected areas of the shell wall can be repaired by weld overlay using approved written weld overlay

procedures. Following repair, the repaired areas shall be examined in accordance with the original inspection requirements and acceptance criteria.

Following thickness testing, the cask cavity weldment, which is the NAC-STC primary containment boundary, shall be hydrostatically tested according to ASME Code, Section III, Subsection NB-6000, as described in Section 8.1.2.3. The cask cavity weldment is dried, the primary containment boundary welds are liquid penetrant examined in accordance with ASME Code, Section V, Article 6, and the welds are accepted in accordance with ASME Code, Section III, Subsection NB-5350. The cask cavity weldment is then helium leak tested to verify that the Containment System Fabrication Verification leak rate is satisfied, as described in Section 8.1.3.

Each of the outer shell sections is rolled from Type 304 stainless steel plate and seam welded longitudinally. The number and length of outer shell sections to be used to achieve the required total outer shell length is optional. The outer shell sections are girth welded to each other and the inside diameter of the "outer shell weldment" is final machined. Longitudinal seam welds in adjacent outer shell sections are offset 180 degrees for girth-welded sections. The outer shell weldment is welded to the cask cavity weldment at the top forging/outer shell interface to form the "body weldment." The preparation, examination, and acceptance procedures for the welds are described in Section 8.1.1 and defined on the License Drawings.

The body weldment is inverted (closure end down) in a pit or other sheltered location in preparation for lead pouring. A temporary dam extension and supports are welded to the open end of the outer shell to permit the full length of the lead shell to be poured and to maintain the outer shell position. "Backing bars" are tack-welded on the inside diameter of the outer shell overlapping the end of the weld prep and on the top surface of the bottom inner forging overlapping the outside diameter of the forging (adjacent to the outside diameter of the inner shell). The backing bars prevent the lead contamination of the welds when the outer shell/bottom outer forging weld and the bottom outer forging/bottom inner forging weld are performed after cask body cooldown following the lead pour. Lead pouring preparations, the pour itself, and the cooldown are performed in accordance with the lead pour requirements and procedures as described in Section 8.4.2.

Following cooldown, the cask may be moved to a location that is more suitable for the fabrication activities that are to follow. The temporary dam extension and supports at the open end of the outer shell are removed and the lead is machined to its final configuration, including facing off the backing bars to ensure that no lead remains on the weld side of the backing bars.

The bottom outer forging is welded to the outer shell and to the bottom inner forging with the backing bars preventing lead contamination of the welds. The weld examination and acceptance criteria are described in Section 8.1.1 and defined on the License Drawings. The NS-4-FR neutron shield material is installed in the bottom forging of the NAC-STC. The NS-4-FR is machined to obtain the specified 2-inch thickness and to provide a groove around the outside diameter. A backing bar is tack-welded on the inside diameter of the bottom outer forging in the groove in the NS-4-FR and flush with its surface. The bottom plate is positioned and welded to the bottom outer forging. The weld examination and acceptance criteria are described in Section 8.1.1 and are defined on the License Drawings.

The outside diameter of the outer shell is then machined to the specified final dimensions. If required to achieve dimensional compliance with the License Drawings, additional localized machining of the inner shell will be performed. Remachined areas of the inner shell shall be re-examined by ultrasonic testing to confirm that the minimum thickness of 1.46 inches (37.1 mm) is maintained. Upon completion of final machining and prior to removal from the machine, the dimensional inspection of the inside diameter and cylindricity of the cavity shall be performed. Using inside micrometers, the inside diameter at 0, 45, 90 and 135 degree radial locations shall be measured. This measurement shall be repeated at a minimum of 6 axial locations through the bore of the inner shell. Using a dial indicator or the electronic measuring system on the machine, a "sweep" of the entire length of the bore at the same radial locations measured with the inside micrometers and also a "sweep" of the diameter at the same axial locations will be performed. The combination of these two inspections will demonstrate the actual diameter and cylindricity of the inner shell bore. Calibrated inspection equipment and approved written procedures will be used to perform the final dimensional inspections.

The Type 17-4 PH stainless steel lifting trunnions are welded to the top forging. The Type 17-4 PH stainless steel rotation trunnion recesses are welded to the outer shell at its juncture with the bottom outer forging. Both the lifting trunnion and rotation trunnion recess weld surfaces are prepared with a minimum 0.25-inch thick overlay of Inconel. The shear ring and the neutron shield upper end plate are welded to the top forging. The weld examination and acceptance criteria are described in Section 8.1.1, and are defined on the License Drawings.

The explosively-bonded stainless steel/copper (SS/Cu) heat transfer fins extending through the neutron shield are welded (only the stainless steel is welded) to the upper end plate and to the outer shell. Following liquid penetrant examination of the fin to outer shell welds, the 24 neutron

shield shell plates are prepared for installation and 1/8-inch thick expansion foam is applied to the interior surface using approved adhesive in accordance with the License Drawings. The neutron shield shell plates are individually positioned and welded to the stainless steel extended tip of the SS/Cu fins. These closure welds are then examined and accepted in accordance with the requirements of the License Drawings. The cask is then placed in the inverted position (closure end down). Following an installation procedure that has been approved by NAC and by the material supplier, the NS-4-FR neutron shield material is installed by pouring into each of the 24 regions between the fins in the NAC-STC neutron shield cavity. After the NS-4-FR has hardened, expansion foam (Section 4.5.3) is installed in the open end of the neutron shield. The inside and outside diametrical (curved) surfaces of the expansion foam are covered by a protective thermal insulation material (Fiberfrax, see Section 4.5.4). The 24 sections of the neutron shield bottom end plate are each positioned and welded to the outer shell, the fins, the neutron shield shell, and to each other. All of the neutron shield and fin welds are liquid penetrant examined and accepted in accordance with the License Drawings. The neutron shield tank is leak tested using the pneumatic bubble method to verify shell integrity.

The Type 17-4 PH stainless steel outer lid forging and the Type 304 stainless steel inner lid forging are machined to the specified final dimensions. The NS-4-FR neutron shield material is installed in the top of the inner lid following an installation procedure that has been approved by NAC and by the material supplier. The exposed surface of the NS-4-FR is machined to obtain the specified 2-inch thickness and the coverplate is welded to the inner lid body. The weld examination and acceptance are in accordance with the requirements of the License Drawings. The top surface of the inner lid is then final machined.

The remaining fabrication details (including the installation of the drain line) are then completed.

Following machining of the structural steel support disks and the aluminum heat transfer disks, the components will be individually inspected for dimensional compliance to the License Drawings to ensure that each disk meets the stated tolerances. The diameter of each disk is measured using a calibrated external micrometer. The openings in each disk are inspected using a calibrated three coordinate measurement machine. The machining center may also be used for these inspections if previously qualified and calibrated. In the case of the diametral tolerances of the disks, the inspections are performed at $65 \pm 5^{\circ}\text{F}$ ($18 \pm 3^{\circ}\text{C}$) or else thermal expansion corrections will be addressed during the inspection process.

The separately fabricated and assembled fuel basket is then inserted into the cask body by carefully guiding the pre-assembled basket into the cask cavity. The acceptance tests described in Section 8.1, not previously completed during fabrication, are performed and the completed NAC-STC is prepared for delivery.

8.4.2 Description of Lead Pour Procedures

This section describes the general requirements and the procedure that applies to the pouring of the lead in the annulus formed by the inner and outer shells of the NAC-STC cask body. The lead annulus provides the primary radial gamma shielding in the cask body and is subjected to a gamma scan test to verify its shielding integrity. The description that follows includes the pre-pour preparations, the pouring of the molten lead in the annulus between the inner and outer shells of the NAC-STC, and the post-pour controlled cooldown of the cask.

8.4.2.1 Preparation for Lead Pour

The following activities must be completed in preparation for pouring of the lead in the NAC-STC cask body:

1. Temporary stiffener bars/rings are installed both inside and outside of the body weldment at intermittent locations along the cask length. The stiffeners support the inner and outer shells during the lead pour and cooldown in order to maintain the specified dimensions of the lead annulus. The stiffeners are removed after the cooldown operation is completed.
2. A minimum of 12 pairs of thermocouples are used to monitor the heating and cooling cycle of the inner and outer shells. Each pair of thermocouples is positioned at approximately the same radial and axial location, one on the inside diameter of the inner shell and one on the outside diameter of the outer shell.
3. Electric heaters are installed in the cask cavity for use in heating the inner shell.
4. The body weldment (Section 8.4.1) of the NAC-STC is inverted and supported in a stable, vertical position in a "pit" or within a windbreak structure to provide a basically draft-free operations area.

5. An auxiliary dam extension and supports are welded to the open end of the outer shell. The extension and supports permit the full length of the lead shell to be poured in one operation while maintaining the annulus spacing at the open end of the outer shell.
6. A minimum of 20 gas heating/water cooling rings are installed around the outside of the body weldment for use in heating, and later in cooling, the outer shell. Gas torches are provided for heating the outside surface of the bottom inner forging.
7. The body weldment surfaces, especially the lead annulus, are checked for dimensional accuracy to ensure that the required spacing has been maintained and for cleanliness to ensure that no foreign materials are present.
8. The general arrangement of the equipment for the lead pour operation is shown in Figure 8.4-1.

8.4.2.2 Lead Pour Operations

The requirements and activities that must be completed during the pouring of the lead in the NAC-STC cask body are:

1. The lead material certification is checked to ensure that it conforms to the requirements of the American Society of Testing Materials (ASTM) B29, Chemical Copper Grade - 99.90 percent pure.
2. Approximately 60,000 pounds of lead is placed in appropriate size kettles and melted. During the lead pouring operations the temperature of the molten lead is maintained between 650°F (343°C) and 750°F (399°C).
3. At the same time that the lead is being melted, the NAC-STC body weldment is simultaneously heated using both the electric heaters on the interior and the gas heating rings on the exterior. The body weldment will be heated in a steady and uniform manner at a rate not exceeding 125°F/hour (52°C/hour). Gas torches are used to heat the exterior of the bottom inner forging. The surface temperature of the body weldment is never permitted to exceed 800°F (427°C). The temperature of the entire

body weldment is maintained between 640°F (338°C) and 740°F (393°C) throughout the lead pour operations.

4. The lead pour is initiated immediately after the temperatures of the lead and the body weldment are stabilized in the ranges previously specified. The actual pouring of the lead is completed without interruption and in as short a period of time as possible. During the lead pour the bottom end of the filler-tube is kept below the surface of the molten lead to preclude the formation of voids in the lead.
5. The lead is poured to a level that is sufficient to ensure that dross removal and contraction during solidification do not reduce the finished surface below the required level. A long steel rod inserted into the molten lead annulus is used to ensure that no solidification has begun anywhere in the volume of molten lead.

8.4.2.3 Cooldown Following Lead Pour

The procedures and requirements that must be completed during cooldown of the NAC-STC body weldment following completion of the lead pour are as follows:

1. Cooldown is initiated by turning off the electrical heater (interior) and the gas heating/water cooling ring (exterior) at the lowest end of the cask (in the as-poured position). The gas heating/water cooling ring is then used to facilitate and control cooling by spraying water on the exterior surface of the cask. As cooldown proceeds, the heaters and rings upward along the cask are successively turned off and the cooling water spray is turned on from each ring.
2. The cooldown process is temperature controlled to maintain approximately uniform solidification conditions across the thickness and around the circumference of the annulus.
3. The cooldown rate is held steady and uniform at a rate not to exceed 125°F/hour (52°C/hour) and the temperature differential between the inside shell and the outside shell is not allowed to exceed 100°F (38°C). Once the inner and outer shell temperatures have cooled to 150°F (66°C), it is no longer necessary to control the cooldown rate.

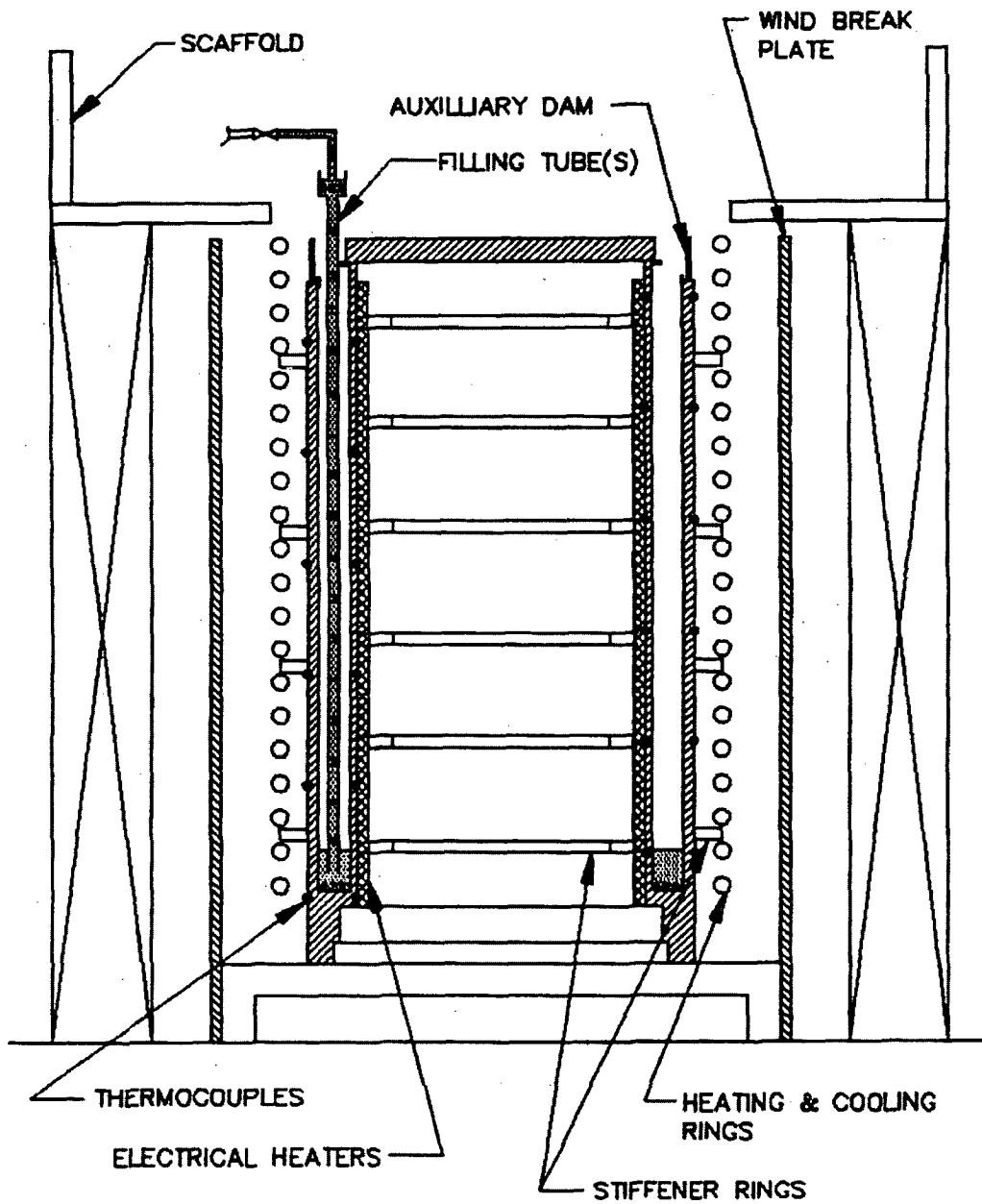
4. The solidification level in the lead annulus is checked with the aid of a long steel rod. The maximum difference in the elevation of the solidified lead between the inside surface of the outer shell and the outside surface of the inner shell is not permitted to exceed 2 inches (51 mm).
5. Dross is skimmed off the top of the lead while maintaining the molten head throughout the cooldown process.

8.4.2.4 Lead Pour Documentation

The following data is included in the Data Package for the Lead Pour Operation:

1. Certificate of Chemical Analysis of the lead.
2. Heating and cooling charts showing elapsed time and temperatures.
3. Location, time and temperature for readings taken with a handheld pyrometer or other temperature reading device.
4. Difference in solidification elevations when checking at the inside surface of the outer shell and the outside surface of the inner shell.

Figure 8.4-1 Arrangement of Lead Pour Equipment



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9.0 **REFERENCES**

10 CFR 20, "Standards for Protection Against Radiation," Part 20, Title 10 of the Code of Federal Regulations.

10 CFR 71, "Packaging and Transportation of Radioactive Materials," Part 71, Title 10 of the Code of Federal Regulations.

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