Nuclear Plant: IP3 I JAF PAGE 1 OF 10

CALC. NO. <u>IP3-CALC-R</u> CALCULATION IS: PRE	<u>C-01718</u> LIMINARY <u>X</u> FII	F NAL	REVISION	_0
PREPARER: CHECKER: (DESIGN) VERIFIED/NA APPROVED:	NAME F. Martsen L. Cerra PHI レ P C X IE K. Eslinger	SIGNATURE Tred III Zelano Co Min Co Willy In	Marting DEC2	DATE 2/9/96 2/9/96 2/9/96 2/9/96 13 1000
ORIGINATOR: SYSTEM NO./NAME	<u>NYPA</u> ⊠ OR <u>OTH</u> RC-MOV-536 Pres	i <u>ER</u> □ sure Locking Analysi FOR INF	S ORMA ONLY	TION
QA CATEGORY:I MODIFICATION NO./TAS	_DISCIPLINE: <u>N</u> K NO. <u>GL-9!</u>	<u>AOV</u> STRUC 5-07 DBD RE	TURE:	N/A N/A
DESIGN BASIS / ASSU Valve Factor = .5, De ALSO SEE P9 4 SUMMARY / CONCLUS Actuator is capable of	MPTION graded Voltage, St SIONS operating under po:	tem Friction Coefficie stulated conditions.	nt = .2	
THIS CALC SUPERSEDES OR VOIDS CALC. NO. N/A				
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COMPONENTS

MAJOR EQUIPMENT	PIPE NO.	VALVE NO.	SUPT. NO	INST. NO.	PENE. NO.
		RC-MOV-536			
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· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·				

RELATED DOCUMENTS

IP3-CALC-RC-01717	
IP3-RPT-MULT-01763	
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RELATED DRAWINGS

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SECURITY: (Y/N) ____

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COMPUTER PRINTOUT: (Y/N)

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L OBJECTIVES:

To determine the required thrust to open the flexible wedge gate valve, RC-MOV-536, under postulated pressure locking conditions and stem thermal growth, and the actuator capability margin for developing that thrust.

II. METHODOLOGY:

The methodology utilized to determine the pressure locking forces is that developed by Commonwealth Edison and presented in Reference 1. The method utilized to determine the stem thermal growth load is based on Commonwealth Edison methodology presented in Reference 2.

Under pressure locking conditions the internal valve pressure forces the two disk halves against the opposing seats resulting in an additional seat contact force. That contact force is a function of the stiffness parameters of the disk plates and central hub. In this methodology these stiffness parameters are approximated by treating the disk plates as uniform circular flat plates with a central hub using the equations from Reference 3 to account for disk plate bending and shear deformation and hub stretch. The stem force required to overcome this additional internal seat contact force is a function of the seat friction and seat angle.

Due to the seat angle, the internal pressure acting on the areas of the disk halves enclosed by the seat contact circumference projected normal to the stem results in a force component on the disk adding to the opening thrust requirement.

From the static condition closing thrust there is a residual seat force which together with the seat angle and seat friction results in a static unwedging force.

For valves that are stroked closed to isolate hot fluid, the portion of stem that is exposed to ambient conditions when open heats up when inserted and expands. For SMB type actuators with self-locking gearing and stem threads, this expansion is restrained. This results in an additional wedging load which must be overcome during unseating.

The stem ejection load assists opening and the packing friction load opposes it.

The above force components are summed to obtain the required opening thrust.

The standard Limitorque methodology with the conservative stem friction coefficient = 0.2 is used to determine the actuator capability.

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III. ASSUMPTIONS:

1. The valve disk is assumed to act as two ideal disks connected by a hub. The equations in Reference 3 are assumed to conservatively model the actual load due to internal pressure. This assumption is considered conservative since the large fillets connecting the hub to the disks make the wedge stiffer than modeled, and expansion of the valve body along the pipe axis due to the internal pressure is neglected. These two effects would reduce the valve internal pressure induced seat contact force and associated required stem thrust.

2. The coefficient of friction at the upstream and downstream seats are assumed to be equal and the same under pressure locking conditions and DP conditions. This assumption is considered to be justified based on ComEd's bench marking of the methodology against ComEd and EPRI pressure locking test data.

3. The Reference 2 methodology assumes that the stiffness of the valve/actuator assembly can be accurately determined from the static diagnostic thrust measurements. This is a reasonable assumption and is considered to be as accurate as the results of an extremely detailed finite element analysis. The stiffness determined in this fashion may be underestimated since it neglects the contribution from the spring pack. Based on preliminary Commonwealth Edison of this effect it is not expected to affect the stiffness more than 20 to 30%. To conservatively accommodate this uncertainty the valve assembly stiffness derived from the test will be increased 100%.

4. The Reference 2 methodology also assumes that the motor speed remains constant during seating. Based on Com. Ed and EPRI testing this is a reasonable assumption for AC motors. The motor RPM utilized in the calculation will be the Limitorque recommended value under load. For these 1800RPM motors that is 1700RPM. The lower number results in a higher stiffness which is conservative.

5. The Reference 2 method assumes that the only portion of stem that undergoes significant thermal growth is the portion inserted into the valve. This is a reasonable assumption as long as the differential temperature is based on the maximum bulk temperature of the fluid in the valve minus the normal ambient temperature.

The Reference 2 methodology is still under review by the Westinghouse Owner's Group (WOG) PLTB Task Team and has not yet been formally issued for utility use. Based on this this calculation will be considered preliminary pending WOG acceptance. The use of this methodology in the interim meets the NRC expectations for use of the "best available information".

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IV. REFERENCES:

- Westinghouse Letter ESBU/WOG-96-022, Attachment 2, "User's Guide For PRESLOCK, A Gate Valve Pressure Locking Analysis Program Using The Commonwealth Edison Model" Rev.0, January 2, 1996.
- 2. Westinghouse Letter ESBU/WOG-96-022, Attachment 4, "User's Guide For STEMGROW, A Gate Valve Thermal Binding Analysis Program Using The Commonwealth Edison Model" Rev.0, December 29, 1996.

3. Sixth Edition of Roark's Formulas for Stress and Strain.

- 4. B&W Doc. 32-1206515-02.
- 5. IP3 Emergency Operating Procedure E-3 Step 12
- 6. Velan Calculation Report DC-124, Rev. 0, June 22, 1993.
- 7. Velan FAX from M. Pang (Velan) to F. Martsen (NYPA), Nov. 29, 1995.
- 8. As-left static test per IP3 Work Request No. 94-03224-01
- 9. NYPA Calculation IP3-CALC-RCS-00978, Rev. 2.
- 10. EPRI/NMAC Report NP-6660-D, March 1990.
- 11. B&W Doc. 51-1224659-00
- 12. SA-1a Product Data Sheet for Armco A-286, Armco Steel Corp.

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V. OPENING FORCE ANALYSIS:

V.1 INPUTS:

Bonnet Pressure	$\mathbf{P}_{\text{bonnet}} = 2510 \cdot \mathbf{psi}$	Reference 4.
Upstream Pressure	P up := 350 psi	Reference 5(Min RCS for . bleed & feed
Downstream Pressure	P down = 0 psi	Assumed (Conservative).
Disk Thickness	t := 0.633·in	Reference 6.
Seat Radius	a := 1.249·in	Reference 6.
Hub Radius	b := 0.915 in	Reference 6.
Half Hub Length	L = 0.367 in (1.0 - 0.633)	Reference 7. (Attached)
Seat Angle	theta = 5 deg	Reference 6.
Poisson's Ratio (disk)	v := .3	Typical of Stainless Steel
Mod. of Elast. (disk)	$\mathbf{E} := 29 \cdot \mathbf{10^6} \cdot \mathbf{psi}$	Typical of Stainless Steel
Static Pullout Force (Nominal Measured)	$F_{po} = 3277 \cdot lbf$	Reference 8.
Diagnostic Error % (SRSS of 10%&2%)	e := 10.2	Reference 9.
Open Valve Factor	VF := .5	Reference 9.
Stem Diameter	$D_{\text{stem}} = 1.125 \cdot \text{in}$	Reference 6.
		Assumed for Press Doghouse
Ambient Temp	T _A = 120	Conserv since max normal is 140F
Max Vlv Temp	T _V ^{:= 669}	Reference 11.
Stroke Length	L _S = 2.6875 in	Reference 6.
Motor Speed	$\mathbf{RPM} := \frac{1700}{\min}$	Reference 9. (Nominal is 1800) (Conservative)

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	Thrust at Seating	F _{HSC} := 960·lbf	Reference 8	
	Thrust at CST	F CST := 14322.lbf	Reference 8	· .
	Unseating thrust	F UW = 2316-lbf	Reference 8	·
	Total Thrust(closing)	F MAX = 15226 lbf	Reference 8	
	Stem Coeff. of Expansi	s. $\alpha := 9.8 \cdot 10^{-6}$	Reference 12	
	Acuator Overall Ratio	OAR = 46.8	Reference 9	·
,	Stem Lead	S _L := 0.2 · in	Reference 9	•
	Time at Seating	t _{hsc} := 20.808 · sec	Reference 8	· · ·
	Time at CST	t _{cst} := 21.618 sec	Reference 8	

V.2 PRESSURE FORCE CALCULATIONS

Coefficient of friction between disk and seat: (Reference 10) $mu := VF \cdot \frac{\cos(\text{theta})}{1 - VF \cdot \sin(\text{theta})}$ mu = 0.52079

Average DP across disks:

DPavg = P bonnet
$$-\frac{P_{up} + P_{down}}{2}$$

Disk Stiffness Constant: (Reference 3, Table 24)

$$D := \frac{E \cdot (t)^3}{12 \cdot (1 - v^2)}$$
$$G := \frac{E}{2 \cdot (1 + v)}$$

DPavg = $2.335 \cdot 10^3$ ·psi

 $D = 6.73576 \cdot 10^5$ ·lbf in

$$G = 1.11538 \cdot 10^7 \cdot psi$$

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Geometry Factors: (Reference 3, Table 24)

$$C_{2} := \frac{1}{4} \cdot \left[1 - \left(\frac{b}{a}\right)^{2} \cdot \left(1 + 2 \cdot \ln\left(\frac{a}{b}\right)\right) \right]$$

$$C_{3} := \frac{b}{4 \cdot a} \cdot \left[\left[\left(\frac{b}{a}\right)^{2} + 1 \right] \cdot \ln\left(\frac{a}{b}\right) + \left(\frac{b}{a}\right)^{2} - 1 \right]$$

$$C_{3} := 0.00272$$

$$C_{3} := \frac{1}{2} \cdot \left[1 + v + (1 - v) \cdot \left(\frac{b}{a}\right)^{2} \right]$$

$$C_{3} := 0.00272$$

$$L_{9} := \frac{a}{a} \cdot \left[\frac{1+v}{2} \cdot \ln\left(\frac{a}{a}\right) + \frac{1-v}{4} \cdot \left[1 - \left(\frac{a}{a}\right)^{2} \right] \right] \qquad L_{9} = 0$$

$$L_{11} := \frac{1}{64} \cdot \left[1 + 4 \cdot \left(\frac{b}{a}\right)^{2} - 5 \cdot \left(\frac{b}{a}\right)^{4} - 4 \cdot \left(\frac{b}{a}\right)^{2} \cdot \left[2 + \left(\frac{b}{a}\right)^{2} \right] \cdot \ln\left(\frac{a}{b}\right) \right] \qquad L_{11} = 1.88551 \cdot 10^{-10}$$

$$L_{17} := \frac{1}{4} \cdot \left[1 - \frac{1 - v}{4} \cdot \left[1 - \left(\frac{b}{a} \right)^4 \right] - \left(\frac{b}{a} \right)^2 \cdot \left[1 + (1 + v) \cdot \ln \left(\frac{a}{b} \right) \right] \right] \qquad L_{17} = 0.03041$$

Moment (Reference 3, Table 24, Case 2L) $M_{rb} := \frac{-DPavg \cdot a^2}{C_8} \cdot \left[\frac{C_9}{2 \cdot a \cdot b} \cdot (a^2 - b^2) - L_{17} \right]$ $M_{rb} := -153.18434 \cdot lbf$ $Q_b := \frac{DPavg}{2 \cdot b} \cdot (a^2 - b^2)$ $Q_b = 922.23058 \cdot \frac{lbf}{in}$



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Deflection due to pressure and bending: (Reference 3, Table 24, Case 2L)

$$y_{bq} = M_{rb} \frac{a^2}{D} C_2 + Q_b \frac{a^3}{D} C_3 - \frac{DPavg a^4}{D} L_{11}$$
 $y_{bq} = -5.80032 \cdot 10^{-6} \cdot in^{-6}$

Deflection due to pressure and shear stress: (Reference 3, Table 25, Case 2L)

 $\mathbf{K}_{\mathbf{sa}} := -0.3 \cdot \left[2 \cdot \ln \left(\frac{\mathbf{a}}{\mathbf{b}} \right) - 1 + \left(\frac{\mathbf{b}}{\mathbf{a}} \right)^2 \right]$ $K_{sa} = -0.04771$ $y_{sq} := \frac{K_{sa} \cdot DPavg \cdot a^2}{t \cdot G}$ $y_{sq} = -2.46143 \cdot 10^{-5}$ ·in

Deflection due to hub stretch (from center of hub to disk):

P force :=
$$3.1416 \cdot (a^2 - b^2) \cdot DPavg$$

P force = $5.30202 \cdot 10^3 \cdot lbf$
y stretch := $\frac{P \text{ force}}{3.1416 \cdot b^2 \cdot (1 \cdot E)}$
y stretch = $2.55104 \cdot 10^{-5} \cdot i$

$$y_{\text{stretch}} = 2.55104 \cdot 10^{-5} \cdot \text{in}$$

Total Deflection due to pressure forces:

 $y_q := y_{bq} + y_{sq} - y_{stretch}$

Deflection due to seat contact force and shear stress (per lbf/in.): (Reference 3, Table 25, Case 1L)

$$\mathbf{y}_{\mathbf{SW}} := -\left[\frac{1.2 \cdot \left(\frac{\mathbf{a}}{\mathbf{a}}\right) \cdot \ln\left(\frac{\mathbf{a}}{\mathbf{b}}\right) \cdot \mathbf{a}}{\mathbf{t} \cdot \mathbf{G}}\right]$$

(per lbf/in)

 $y_{q} = -5.5925 \cdot 10^{-5} \cdot in$

 $\mathbf{y}_{sw} = -6.60571 \cdot 10^{-8} \cdot \frac{\text{in}}{(\underline{lbf})}$

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 $y_{compr} = 3.77587 \cdot 10^{-8}$ ·

 $\int in /$

 $\frac{1 \text{bf}}{1 \text{in}}$

Deflection due to seat contact force and bending (per lbf/in.): (Reference 3, Table 24, Case 1L)

$$y_{bw} = -\left(\frac{a^{3}}{D}\right) \cdot \left[\left(\frac{C_{2}}{C_{8}}\right) \cdot \left[\left(\frac{a \cdot C_{9}}{b}\right) - L_{9}\right] - \left[\left(\frac{a}{b}\right) \cdot C_{3}\right] + L_{3}\right] \quad y_{bw} = -2.08804 \cdot 10^{-8} \cdot \frac{in}{\left(\frac{lbf}{b}\right)}$$

(per lbf/in)

Deflection due to hub compression (per lbf/in), (from center of hub to disk):

$$y_{compr} := \frac{2 \cdot a \cdot \pi}{3.1416 \cdot b^2} \cdot \frac{L}{(1 \cdot E)}$$

(per lbf/in)

Total deflection due to seat contact force (per lbf/in.):

 $y_{w} = y_{bw} + y_{sw} - y_{compr}$ $y_{w} = -1.24696 \cdot 10^{-7} \cdot \frac{\text{in}}{\left(\frac{\text{lbf}}{\text{in}}\right)}$ (per lbf/in)

Seat Contact Force for which deflection is equal previously calculated deflection from pressure forces:

 $F_{s} = 2 \cdot \pi \cdot a \cdot \frac{y_{q}}{y_{w}}$ $F_{s} = 3.51961 \cdot 10^{3} \cdot lbf$

V.3 UNSEATING FORCES

F_{rectine} is included in measured static pullout Force

$$F_{pomax} = \frac{F_{po}}{1 - \frac{e}{100}}$$

$$F_{\text{pomax}} = 3.64922 \cdot 10^3 \cdot \text{lbf}$$

Speed := RPM
$$\frac{1}{60 \cdot \frac{\sec}{\min}} \cdot \frac{SL}{OAR}$$

Rate = $\frac{F_{CST} - F_{HSC}}{t_{cst} - t_{hsc}}$

Speed =
$$0.12108 \cdot \sec^{-1}$$
 ·in

Rate = $1.64963 \cdot 10^4 \cdot \sec^{-1} \cdot lbf$

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 $K := \frac{2 \cdot \text{Rate}}{\text{Speed}} \qquad K = 3.26976 \cdot 10^6 \cdot \text{ft}^{-1} \cdot \text{lbf} \qquad \text{(Doubled for conservatism)}$

 $F_{\text{thermal}} = K \cdot \alpha \cdot L_{S} \cdot \begin{pmatrix} T_{V} - T_{A} \end{pmatrix}$ $F_{\text{thermal}} = 3.93987 \cdot 10^{3} \cdot lbf$ $F_{\text{thermunwdg}} = \frac{F_{UW}}{F_{MAX} - F_{HSC}} \cdot F_{\text{thermal}}$ $F_{\text{thermunwdg}} = 639.61385 \cdot lbf$

 $F_{\text{piston}} = \frac{\pi}{4} \cdot D_{\text{stem}}^2 \cdot P_{\text{bonnet}}$ $F_{\text{piston}} = 2.49499 \cdot 10^3 \cdot 10^5$

 $F_{vert} := \pi \cdot a^{2} \cdot sin(theta) \cdot (2 \cdot P_{bonnet} - P_{up} - P_{down}) F_{vert} = 1.99475 \cdot 10^{3} \cdot lbf$ $F_{preslock} := 2 \cdot F_{s} \cdot (mu \cdot cos(theta) - sin(theta)) F_{preslock} = 3.03852 \cdot 10^{3} \cdot lbf$ $F_{total} := -F_{piston} + F_{vert} + F_{preslock} + F_{pomax} + F_{thermunwdg}$

 $F_{total} = 6.82711 \cdot 10^3$ ·lbf

VI. CONCLUSION

Per Reference 9 the calculated actuator thrust capability at degraded voltage and elevated temperature and with a conservative stem friction coefficient of 0.2 is: MCUVo = 8828lbf. Since this is greater than the above calculated force, $F_{total} = 6827$ lbf, the capability of this MOV to open under the postulated pressure locking conditions combined with the stem thermal growth load is demonstrated. The margin is ((8828 - 6827)/(6827))x100% = 29.3%.



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Pressure Locking Evaluation of RI-MOV-536

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Verification of:

Document Title:

Document Number:

Subject:

- CALC - RC- 01718 Pressure Locking Evaluation of RC-MOV-

Modification/Task Number (if applicable):

QA Category:

CAT I

	Review Required	Discipline	Review Complete (initial of reviewer)
		ELECTRICAL	
		MECHANICAL	
		INSTRUMENT & CONTROL	
•	-	CIVIL/ STRUCTURAL	· .
		FIRE PROTECTION	
	· .	SIMULATOR	





New York Power Authority 図 · IP3 **DESIGN VERIFICATION CHECKLIST** page 1 of 4 **IDENTIFICATION:** DISCIPLINE: Document Title: <u>fressure Looking Evaturations of</u> RC-MOV-536 **18C** [] MECH [] Fire Protect Doc. Number: 1P3- CALC-RC-01718 Doc. Revision: [] C/S Simulator 0 П Other. CAT 1 (specify) QA Category: . **METHOD OF VERIFICATION:** [] **Design Review** Alternate Calculations D. 11 **Qualification Test** PHILIP C XIE, SYSTEM ENGINEER. -6017 Selected Verifier: 4 Design Verification Questionnaire All questions shall be explained in the space provided. 1. Were the inputs correct and incorporated into the design? data Explanation: AII Yes have pii Aina been reterence. Contirmed Are the physical and functional characteristics of the proposed design within the approved design basis of the 2. system(s) structure(s) or component(s)? Explanation: З. Does the proposed design incorporate license Commitments? Explanation: N/A Are assumptions necessary to perform the design activity adequately described and reasonable: Where 4. necessary, are the assumptions identified for subsequent reverifications when the detailed design activities are completed? Explanation: Yes Assume times ìn Are the appropriate quality and quality assurance requirements specified? e.g., safety classification? 5. **Explanation:** this MO atety les Are the applicable codes, standards and regulatory requirements including issue and addenda property 6. identified and are their requirements for design met? Explanation:

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*	Design Verification Questionnaire All questions shall be explained in the space provided
7.	Have applicable construction and operating experience been considered?
Explana	ition: Yes This columbritism supports the review for PLTB under GL 95-07. that review includes considerations of operations
	exporional.
8.	Have the design interface requirements for mechanical, electrical/I&C, and civil/structural angineering be satisfied?
Explana	ition:
	N/A
9.	Was the appropriate design method used?
Explana	ition: Yes this calculation uses the standard inductry methods
	double - sice date value.
10.	Is the output reasonable compared to inputs?
Explana	ition:
	Yes
11.	Are the specified parts, equipment and processes properly suited for the fire protection Appendix R, QA, an EQ classifications required for the application?
Explana	
	N/A
12.	Are the specified materials compatible with each other and the design environmental conditions to which the material will be exposed?
Explana	ition:
	N/A
13.	Have personnel requirements and limitations for maintenance, testing, and inspection been satisfied?
Explana	ition:
	N/A
14.	Are accessibility, maintenance, repair, and inservice inspection requirements for the plant including the plan conditions under which these will be performed been considered?
Explana	ition:
	NA
15.	Has adequate accessibility been provided to perform the in-service inspection expected to be required during the elect life?

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DESIGN VERIFICATION

ATTACHMENT 4.2



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Design Verification Guestionnaire All questions shall be explained in the space provided Has the design property considered radiation exposure to the public and plant personnel? (ALARA/ cobalt 16. reduction) Explanation: Ν Are the acceptance criteria incorporated in the design documents sufficient to allow verification that design 17. requirements have satisfactorily accomplished? Explanation: the conclusion reached directhy imporme requirements Capability Have adequate pre-operational and subsequent periodic test requirements been appropriately specified? 18. **Explanation:** 19. Are adequate handling, storage, cleaning and shipping requirements specified? Explanation: Are adequate identification requirements specified? 20. Explanation: N Are the conclusions drawn in the Safety Evaluation fully supported by adequate discussion in the test or Safety 21. Evaluation itself? Explanation: Are necessary procedural changes specified, and are responsibilities for such changes clearly delineated? 22. Explanation: • NI Are requirements for record preparation, review, approval, retention, etc., adequately specified? 23. Explanation: This CAT il caludation Documents Contr. Dr agess Have supplemental reviews by other engineering disciplines (seismic, electrical, etc.) been performed on the 24. integrated design package? Explanation: enginal WI 0 engineer calestation reviewell Have the drawings, sketches, calculations, etc., included in the integrated design package been reviewed? 25. **Explanation:** 'eX

DESIGN VERIFICATION



DESIGN VERIFICATION CHECKLIST page 4 of 4

Design Verification Questionnaire All questions shall be explained in the space provided Have reviews been performed to identify any effect on the Check Valve Maintenance Program? 26. **Explanation:** NI A Does the design for check valves meet the intents of INPO SOER 86-03? 27. **Explanation:** N Is the plant reference simulator physical and functional fidelity affected and it's design change been factored into the cost? 28. **Explanation:** Are all references listed (including design calculation/analysis) that were used as part of the design review? 29. **Explanation:** les. **REMARKS/COMMENTS:** 2/10/96 **Design Verification** Complete:

