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**WCAP-16752-NP, Revision 0,**  
**“Indian Point Unit 2 Heatup and Cooldown Limit Curves for Normal Operation”**  
**Westinghouse Electric Co, January 2008**

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January 2008

# **Indian Point Unit 2 Heatup and Cooldown Limit Curves for Normal Operation**



WCAP-16752-NP

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**Indian Point Unit 2  
Heatup and Cooldown Limit Curves  
for Normal Operation**

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**PREFACE**

This report has been technically reviewed and verified by: F.C. Gift\*

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**RECORD OF REVISION**

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## EXECUTIVE SUMMARY

This report provides the methodology and results of the generation of heatup and cooldown pressure-temperature (PT) limit curves for normal operation of the Indian Point Unit 2 reactor vessel. The PT curves were generated based on the latest available reactor vessel information and updated calculated fluences. The new Indian Point Unit 2 heatup and cooldown pressure-temperature limit curves were generated using the "Axial-Flaw" methodology of the 1998 ASME Code, Section XI through the 2000 Addenda (which allows the use of the  $K_{Ic}$  methodology) and the less restrictive "Circ-Flaw" methodology (both methodologies formally known as ASME Code Cases N-640 and N-588, respectively). The material with the highest adjusted reference temperature (ART) was Intermediate Shell Plate B-2002-3 (Heat Number B4782-1) for the "Axial-Flaw" methodology and circumferential weld wire heat B34009 for the "Circ-Flaw" methodology. The PT limit curves were generated for 29.2 and 48 EFPY using heatup rates of 60 and 100°F/hr and cooldown rates of 0, 20, 40, 60 and 100°F/hr. Lastly, the PT Curves were developed without instrumentation errors. These curves can be found in Figures 5-1 through 5-4.

## 1 INTRODUCTION

Heatup and cooldown limit curves are calculated using the adjusted  $RT_{NDT}$  (reference nil-ductility temperature) corresponding to the limiting beltline region material of the reactor vessel. The adjusted  $RT_{NDT}$  of the limiting material in the core region of the reactor vessel is determined by using the unirradiated reactor vessel material fracture toughness properties, estimating the radiation-induced  $\Delta RT_{NDT}$ , and adding a margin. The unirradiated  $RT_{NDT}$  is designated as the higher of either the drop weight nil-ductility transition temperature (NDTT) or the temperature at which the material exhibits at least 50 ft-lb of impact energy and 35-mil lateral expansion (normal to the major working direction) minus 60°F.

$RT_{NDT}$  increases as the material is exposed to fast-neutron radiation. Therefore, to find the most limiting  $RT_{NDT}$  at any time period in the reactor's life,  $\Delta RT_{NDT}$  due to the radiation exposure associated with that time period must be added to the unirradiated  $RT_{NDT}$  ( $IRT_{NDT}$ ). The extent of the shift in  $RT_{NDT}$  is enhanced by certain chemical elements (such as copper and nickel) present in reactor vessel steels. The Nuclear Regulatory Commission (NRC) has published a method for predicting radiation embrittlement in Regulatory Guide 1.99, Revision 2, "Radiation Embrittlement of Reactor Vessel Materials" [Reference 1]. Regulatory Guide 1.99, Revision 2, is used for the calculation of Adjusted Reference Temperature (ART) values ( $IRT_{NDT} + \Delta RT_{NDT} +$  margins for uncertainties) at the 1/4T and 3/4T locations, where T is the thickness of the vessel at the beltline region measured from the clad/base metal interface.

The heatup and cooldown curves documented in this report were generated using the most limiting ART values and the NRC approved methodology documented in WCAP-14040-NP-A, Revision 4 [Reference 2], "Methodology Used to Develop Cold Overpressure Mitigating System Setpoints and RCS Heatup and Cooldown Limit Curves." Specifically, the heatup and cooldown curves documented in this report were generated using the most limiting ART values, specifically for the "Axial-Flaw" and "Circ-Flaw" methodologies of the 1998 ASME Code, Section XI through the 2000 Addenda which makes use of the  $K_{Ic}$  methodology. The purpose of this report is to present the calculations and the development of the Indian Point Unit 2 heatup and cooldown curves for 29.2 and 48 EFPY. This report documents the calculated ART values and the development of the PT limit curves for normal operation. The PT curves herein were generated without instrumentation errors. The PT curves include the pressure-temperature limits for the vessel flange region per the requirements of 10 CFR Part 50, Appendix G [Reference 3].

## 2 FRACTURE TOUGHNESS PROPERTIES

The fracture-toughness properties of the ferritic materials in the reactor coolant pressure boundary are determined in accordance with the NRC Standard Review Plan [Reference 4]. The beltline material properties of the Indian Point Unit 2 reactor vessel are presented in Table 2-1. Best estimate copper (Cu) and nickel (Ni) weight percent values used to calculate chemistry factors (CF) in accordance with Regulatory Guide 1.99, Revision 2, are provided in Table 2-1. Additionally, surveillance capsule data is available for four capsules (Capsules V, Z, Y and T) already removed from the Indian Point Unit 2 reactor vessel. The fluence data for the surveillance capsules is presented in Table 2-2 and is used to calculate CF values per Position 2.1 of Regulatory Guide 1.99, Revision 2. It should be noted that in addition to Indian Point Unit 2, surveillance weld data from Indian Point Unit 3 and H.B. Robinson Unit 2 was used in the determination of CF. In addition, all the surveillance data has been determined to be credible, with exception of surveillance plate B-2002-2.

The chemistry factors were calculated using Regulatory Guide 1.99 Revision 2, Positions 1.1 and 2.1. Position 1.1 uses the Tables from the Reg. Guide along with the best estimate copper and nickel weight percents. Position 2.1 uses the surveillance capsule data from all capsules withdrawn to date, including those capsules from Indian Point Unit 3 and H.B. Robinson Unit 2. The measured  $\Delta RT_{NDT}$  values for the weld data were adjusted for the temperature difference between differing plants and for chemistry using the ratio procedure given in Position 2.1 of Regulatory Guide 1.99, Revision 2. Table 2-3 contains the  $T_{cold}$  operating temperatures at Indian Point Units 2 and 3 and H.B. Robinson Unit 2. Table 2-4 details the calculation of the surveillance material chemistry factors. A summary of the resulting CF values for all of the vessel and surveillance materials is presented in Table 2-5.

**TABLE 2-1**  
**Summary of the Best Estimate Cu and Ni Weight Percent and Initial RT<sub>NDT</sub> Values for the**  
**Indian Point Unit 2 Reactor Vessel Materials**

| <b>Material Description</b>   | <b>Cu (%)</b> | <b>Ni (%)</b> | <b>Initial RT<sub>NDT</sub><sup>(a)</sup></b> |
|---|---------------|---------------|---|
| Closure Head Flange   | ---           | ---           | 60°F  |
| Vessel Flange   | ---           | ---           | 60°F  |
| Intermediate Shell Plate B-2002-1 <sup>(e)</sup>                                    | 0.19 (0.21)   | 0.65 (0.62)   | 34°F  |
| Intermediate Shell Plate B-2002-2 <sup>(e)</sup>                                    | 0.17 (0.15)   | 0.46 (0.44)   | 21°F  |
| Intermediate Shell Plate B-2002-3 <sup>(e)</sup>                                    | 0.25 (0.20)   | 0.60 (0.59)   | 21°F  |
| Lower Shell Plate B-2003-1  | 0.20          | 0.66          | 20°F  |
| Lower Shell Plate B-2003-2  | 0.19          | 0.48          | -20°F   |
| Intermediate & Lower Shell Longitudinal Weld Seams (Heat # W5214) <sup>(b, d)</sup> | 0.21          | 1.01          | -56°F   |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009) <sup>(c, d)</sup>            | 0.19          | 1.01          | -56°F   |
| Indian Point Unit 2 Surveillance Weld (Heat # W5214) <sup>(b, d)</sup>              | 0.20          | 0.94          | ---   |
| Indian Point Unit 3 Surveillance Weld (Heat # W5214) <sup>(b, d)</sup>              | 0.16          | 1.12          | ---   |
| H.B. Robinson Unit 2 Surveillance Weld (Heat # W5214) <sup>(b, d)</sup>             | 0.32          | 0.66          | ---   |

Notes:

- (a) The Initial RT<sub>NDT</sub> values are measured values, with exception to the weld materials.
- (b) The weld material in the Indian Point Unit 2 surveillance program was made of the same wire and flux as the reactor vessel intermediate shell longitudinal weld seams (Wire Heat No. W5214 RACO3 + Ni200, Flux Type Linde 1092, Flux Lot No. 3600). The lower shell longitudinal weld seam also had the same heat and flux type but different flux lot. Indian Pt. Unit 3 and H.B. Robinson Unit 2 also contain surveillance material of this heat.
- (c) The intermediate to lower shell circ. weld material was made of Wire Heat No. 34B009 RACO3 + Ni200, Flux Type Linde 1092, Flux Lot No. 3708.
- (d) The weld best estimate copper and nickel weight percents were obtained from CE Reports NPSD-1039, Rev. 2 [Reference 5] and/or NPSD-1119, Rev. 1 [Reference 6]. The values from the CE Report NPSD-1119, Rev. 1 for the Indian Point 2 vessel axial and circ. welds match those in the NRC database RVID2. The values were rounded to two decimal points.
- (e) Copper and Nickel Values were obtained from WCAP-12796 [Reference 7], which in turn used Southwest Research Report 17-2108 (Capsule V Analysis). This report calculated a best estimate Copper/Nickel weight percent excluding values that appeared to be outliers. If all data was considered, then the best estimate would match the RVID2 values shown in parenthesis. The data above for the intermediate shell plates are conservative with exception to plate B-2002-1. The RVID2 chemistry for plate B-2002-1 produces a Regulatory Guide 1.99 Position 1.1 chemistry factor of 156.2°F as compared to the chemistry factor of 144°F calculated in Reference 7. Credible surveillance data (See Tables 2-4 & 2-5) is used to provide a Position 2.1 chemistry factor of 114°F. Intermediate shell plate B-2002-3 is more limiting than B-2002-1 even if the highest CF were used for B-2002-1. Values from WCAP-12796 will be used herein.

**TABLE 2-2**  
**Calculated Integrated Neutron Exposure of the Surveillance Capsules @ Indian Point Unit 2, Indian Point Unit 3 and H.B. Robinson Unit 2 [Reference 8, 15, 16]**

| Capsule                     | Fluence  |
|-----------------------------|--|
| <b>Indian Point Unit 2</b>  |  |
| T                           | $2.53 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV)                |
| Y                           | $4.55 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV)                |
| Z                           | $1.02 \times 10^{19}$ n/cm <sup>2</sup> , (E > 1.0 MeV)                |
| V                           | $4.92 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV)                |
| <b>Indian Point Unit 3</b>  |  |
| T                           | $2.63 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| Y                           | $6.92 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| Z                           | $1.04 \times 10^{19}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| X                           | $8.74 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| <b>H.B. Robinson Unit 2</b> |  |
| S                           | $4.79 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| V                           | $5.30 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| T                           | $3.87 \times 10^{19}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |
| X                           | $4.49 \times 10^{18}$ n/cm <sup>2</sup> , (E > 1.0 MeV) <sup>(a)</sup> |

(a) Fluence values have been adjusted to be consistent with the methodology of Regulatory Guide 1.190 [Reference 13]

**TABLE 2-3**  
**Inlet ( $T_{cold}$ ) Operating Temperatures [Reference 8]**

| <b>Indian Point Unit 2</b>           | <b>Indian Point Unit 3</b> | <b>H.B. Robinson Unit 2</b> |
|--------------------------------------|----------------------------|-----------------------------|
| 543°F (Cycle 1)                      | 540°F (Capsule T)          | 547°F (Capsule S)           |
| 543°F (Cycle 2)                      | 540°F (Capsule Y)          | 547°F (Capsule T)           |
| 522.5°F (Cycle 3)                    | 540°F (Capsule Z)          | 547°F (Capsule X)           |
| 522.5°F (Cycle 4)                    | 540°F (Capsule X)          |                             |
| 522.8°F (Cycle 5)                    |                            |                             |
| 522.8°F (Cycle 6)                    |                            |                             |
| 522.8°F (Cycle 7)                    |                            |                             |
| 522.5°F (Cycle 8)                    | ---                        | ---                         |
| <b>528°F (Average)<sup>(a)</sup></b> | <b>540°F (Average)</b>     | <b>547°F (Average)</b>      |

(a) The temperatures listed above are consistent with historical treatment in previous Pressure-Temperature WCAP's, but are slightly conservative compared to measured operating history.

**TABLE 2-4**  
**Calculation of Chemistry Factors using Indian Point Unit 2 Surveillance Capsule Data**

| Material  | Capsule  | Capsule $f^{(a)}$ | FF <sup>(b)</sup> | $\Delta RT_{NDT}^{(c)}$ | FF* $\Delta RT_{NDT}$ | FF <sup>2</sup> |
|---|--|-------------------|-------------------|-------------------------|-----------------------|-----------------|
| Intermediate Shell<br>Plate B-2002-1  | T  | 0.253             | 0.627             | 55.0                    | 34.49                 | 0.393           |
|   | Z  | 1.02              | 1.006             | 125.0                   | 125.75                | 1.012           |
|   | SUM:   |                   |                   |                         | 160.24                | 1.405           |
|   | $CF_{B-2002-1} = \frac{\sum(FF * RT_{NDT})}{\sum(FF^2)} = \frac{(160.24)}{(1.405)} = 114.0^{\circ}F$ |                   |                   |                         |                       |                 |
| Intermediate Shell<br>Plate B-2002-2  | T  | 0.253             | 0.627             | 95.0                    | 59.57                 | 0.393           |
|   | Z  | 1.02              | 1.006             | 120.0                   | 120.72                | 1.012           |
|   | V  | 0.492             | 0.802             | 77.0                    | 61.75                 | 0.643           |
|   | SUM:   |                   |                   |                         | 242.04                | 2.048           |
| $CF_{B-2002-2} = \frac{\sum(FF * RT_{NDT})}{\sum(FF^2)} = \frac{(242.04)}{(2.048)} = 118.2^{\circ}F$            |  |                   |                   |                         |                       |                 |
| Intermediate Shell<br>Plate B-2002-3  | T  | 0.253             | 0.627             | 115.0                   | 72.11                 | 0.393           |
|   | Y  | 0.455             | 0.781             | 145.0                   | 113.25                | 0.610           |
|   | Z  | 1.02              | 1.006             | 180.0                   | 181.08                | 1.012           |
|   | SUM:   |                   |                   |                         | 366.44                | 2.015           |
| $CF_{B-2002-3} = \frac{\sum(FF * RT_{NDT})}{\sum(FF^2)} = \frac{(366.44)}{(2.015)} = 181.9^{\circ}F$            |  |                   |                   |                         |                       |                 |
| Surveillance Weld<br>Material <sup>(d)</sup>  | Y (IP2)  | 0.455             | 0.781             | 208.7 (195)             | 162.9                 | 0.610           |
|   | V (IP2)  | 0.492             | 0.802             | 218.3 (204)             | 175.1                 | 0.643           |
|   | T (IP3)  | 0.263             | 0.637             | 183.2(151.6)            | 116.7                 | 0.405           |
|   | Y (IP3)  | 0.692             | 0.897             | 206.1(172.0)            | 184.8                 | 0.804           |
|   | Z (IP3)  | 1.04              | 1.011             | 270.1 (229.2)           | 273.1                 | 1.022           |
|   | X(IP3)   | .874              | .962              | 229.8 (193.2)           | 221.1                 | 0.926           |
|   | V(HBR2)  | 0.530             | 0.823             | 248.9(209.3)            | 204.7                 | 0.677           |
|   | T(HBR2)  | 3.87              | 1.349             | 334.8 (288.2)           | 451.6                 | 1.820           |
|   | X(HBR2)  | 4.49              | 1.381             | 310.6 (265.9)           | 428.8                 | 1.906           |
|   | SUM:   |                   |                   |                         | 2218.9                | 8.813           |
| $CF_{Surv. Weld} = \frac{\sum(FF * RT_{NDT})}{\sum(FF^2)} = \frac{(2218.9^{\circ}F)}{(8.813)} = 251.8^{\circ}F$ |  |                   |                   |                         |                       |                 |

**Notes:**

- (a)  $f$  = fluence. See Table 2-3, ( $\times 10^{19}$  n/cm<sup>2</sup>,  $E > 1.0$  MeV).
- (b) FF = fluence factor =  $f^{(0.28 + 0.1 \cdot \log f)}$ .
- (c)  $\Delta RT_{NDT}$  values are the measured 30 ft-lb shift values taken from the following documents:
  - Indian Point Unit 2 Plate and Weld... WCAP-12796 (Refers back to the original Southwest Research Institute Report for each capsule.) [Reference 7]
  - Indian Point Unit 3 Weld... WCAP-16251 [Reference 16].
  - H.B. Robinson Unit 2... Letter Report CPL-96-203 [Reference 10]
- (d) Per Table 2 Indian Point Unit 3 operates with an inlet temperature of approximately 540°F, H.B. Robinson Unit 2 operates with an inlet temperature of approximately 547°F, and Indian Point Unit 2 operates with an inlet temperature of approximately 528°F. The measured  $\Delta RT_{NDT}$  values from the Indian Point Unit 3 surveillance program were adjusted by adding 12°F to each measured  $\Delta RT_{NDT}$  and the H.B. Robinson Unit 2 surveillance program values were adjusted by adding 19°F to each measured  $\Delta RT_{NDT}$  value before applying the ratio procedure. The surveillance weld metal  $\Delta RT_{NDT}$  values have been adjusted by a ratio factor of:
  - Ratio IP2 =  $230.2 \div 214.3 = 1.07$  for the Indian Point Unit 2 data.
  - Ratio IP3 =  $230.2 \div 206.2 = 1.12$  for the Indian Point Unit 3 data.
  - Ratio HBR2 =  $230.2 \div 210.7 = 1.09$  for the H.B. Robinson Unit 2 data.
 Refer to Table 2-5 for the longitudinal weld seam CF of 230.2°F.  
 (The pre-adjusted values are in parenthesis.)

**TABLE 2-5**  
**Summary of the Indian Point Unit 2 Reactor Vessel Beltline Material Chemistry Factors**

| <b>Material</b>   | <b>Reg. Guide 1.99, Rev. 2<br/>Position 1.1 CF's</b> | <b>Reg. Guide 1.99, Rev. 2<br/>Position 2.1 CF's</b> |
|---|--|--|
| Intermediate Shell Plate B-2002-1                                       | 144°F  | 114  |
| Intermediate Shell Plate B-2002-2                                       | 115.1°F  | 118.2  |
| Intermediate Shell Plate B-2002-3                                       | 176°F  | 181.9  |
| Lower Shell Plate B-2003-1  | 152°F  | ---  |
| Lower Shell Plate B-2003-2 <sup>(a)</sup>                               | 128.8°F  | ---  |
| Intermediate & Lower Shell<br>Longitudinal Weld Seams<br>(Heat # W5214) | 230.2°F  | 251.8  |
| Intermediate to Lower Shell<br>Girth Weld Seam (Heat # 34B009)          | 220.9°F  | ---  |
| Indian Point Unit 2 Surveillance<br>Weld (Heat # W5214)                 | 214.3°F  | ---  |
| Indian Point Unit 3 Surveillance<br>Weld (Heat # W5214)                 | 206.2°F  | ---  |
| H.B. Robinson Unit 2 Surveillance<br>Weld (Heat # W5214)                | 210.7°F  | ---  |

(a) The 128.8°F CF listed here differs from previous analyses that used an excessively conservative CF of 142°F for this material. If the 142°F CF had been used in this analysis, this material would still not be limiting.



### 3 CRITERIA FOR ALLOWABLE PRESSURE-TEMPERATURE RELATIONSHIPS

#### 3.1 OVERALL APPROACH

The ASME approach for calculating the allowable limit curves for various heatup and cooldown rates specifies that the total stress intensity factor,  $K_I$ , for the combined thermal and pressure stresses at any time during heatup or cooldown cannot be greater than the reference stress intensity factor,  $K_{Ic}$ , for the metal temperature at that time.  $K_{Ic}$  is obtained from the reference fracture toughness curve, defined in the 1998 Edition through the 2000 Addenda of Section XI, Appendix G of the ASME Code [Reference 11]. The  $K_{Ic}$  curve is given by the following equation:

$$K_{Ic} = 33.2 + 20.734 * e^{[0.02(T - RT_{NDT})]} \quad (1)$$

where,

$K_{Ic}$  = reference stress intensity factor as a function of the metal temperature  $T$  and the metal reference nil-ductility temperature  $RT_{NDT}$

This  $K_{Ic}$  curve is based on the lower bound of static critical  $K_I$  values measured as a function of temperature on specimens of SA-533 Grade B Class\_1, SA-508-1, SA-508-2, SA-508-3 steel.

#### 3.2 METHODOLOGY FOR PRESSURE-TEMPERATURE LIMIT CURVE DEVELOPMENT

The governing equation for the heatup-cooldown analysis is defined in Appendix G of the ASME Code as follows:

$$C * K_{Im} + K_{It} < K_{Ic} \quad (2)$$

where,

$K_{Im}$  = stress intensity factor caused by membrane (pressure) stress

$K_{It}$  = stress intensity factor caused by the thermal gradients

$K_{Ic}$  = function of temperature relative to the  $RT_{NDT}$  of the material

$C$  = 2.0 for Level A and Level B service limits

$C$  = 1.5 for hydrostatic and leak test conditions during which the reactor core is not critical

For membrane tension, the corresponding  $K_I$  for the postulated defect is:

$$K_{Im} = M_m \times (pR_i / t) \quad (3)$$

where,  $M_m$  for an inside surface flaw is given by:

$$\begin{aligned} M_m &= 1.85 \text{ for } \sqrt{t} < 2, \\ M_m &= 0.926\sqrt{t} \text{ for } 2 \leq \sqrt{t} \leq 3.464, \\ M_m &= 3.21 \text{ for } \sqrt{t} > 3.464 \end{aligned}$$

Similarly,  $M_m$  for an outside surface flaw is given by:

$$\begin{aligned} M_m &= 1.77 \text{ for } \sqrt{t} < 2, \\ M_m &= 0.893\sqrt{t} \text{ for } 2 \leq \sqrt{t} \leq 3.464, \\ M_m &= 3.09 \text{ for } \sqrt{t} > 3.464 \end{aligned}$$

and

$p$  = internal pressure,  $R_i$  = vessel inner radius, and  $t$  = vessel wall thickness.

For bending stress, the corresponding  $K_I$  for the postulated defect is:

$$K_{Ib} = M_b * \text{Maximum Stress, where } M_b \text{ is two-thirds of } M_m$$

The maximum  $K_I$  produced by radial thermal gradient for the postulated inside surface defect of G-2120 is  $K_{It} = 0.953 \times 10^{-3} \times CR \times t^{2.5}$ , where  $CR$  is the cooldown rate in °F/hr., or for a postulated outside surface defect,  $K_{It} = 0.753 \times 10^{-3} \times HU \times t^{2.5}$ , where  $HU$  is the heatup rate in °F/hr.

The through-wall temperature difference associated with the maximum thermal  $K_I$  can be determined from Fig. G-2214-1. The temperature at any radial distance from the vessel surface can be determined from Fig. G-2214-2 for the maximum thermal  $K_I$ .

- (a) The maximum thermal  $K_I$  relationship and the temperature relationship in Fig. G-2214-1 are applicable only for the conditions given in G-2214.3(a)(1) and (2).
- (b) Alternatively, the  $K_I$  for radial thermal gradient can be calculated for any thermal stress distribution and at any specified time during cooldown for a ¼-thickness inside surface defect using the relationship:

$$K_{It} = (1.0359C_0 + 0.6322C_1 + 0.4753C_2 + 0.3855C_3) * \sqrt{\pi a} \quad (4)$$

or similarly,  $K_{It}$  during heatup for a ¼-thickness outside surface defect using the relationship:

$$K_{It} = (1.043C_0 + 0.630C_1 + 0.481C_2 + 0.401C_3) * \sqrt{\pi a} \quad (5)$$

where the coefficients  $C_0$ ,  $C_1$ ,  $C_2$  and  $C_3$  are determined from the thermal stress distribution at any specified time during the heatup or cooldown using the form:

$$\sigma(x) = C_0 + C_1(x/a) + C_2(x/a)^2 + C_3(x/a)^3 \quad (6)$$

and  $x$  is a variable that represents the radial distance from the appropriate (i.e., inside or outside) surface to any point on the crack front and  $a$  is the maximum crack depth.

Note, that equations 3, 4 and 5 were implemented in the OPERLIM computer code, which is the program used to generate the pressure-temperature (PT) limit curves. No other changes were made to the OPERLIM computer code with regard to PT calculation methodology. Therefore, the PT curve methodology is unchanged from that described in WCAP-14040-NP-A, "Methodology used to Develop Cold Overpressure Mitigating System Setpoints and RCS Heatup and Cooldown Limit Curves" [Reference 2] Section 2.6 (equations 2.6.2-4 and 2.6.3-1) with the exceptions just described above.

At any time during the heatup or cooldown transient,  $K_{Ic}$  is determined by the metal temperature at the tip of a postulated flaw at the 1/4T and 3/4T location, the appropriate value for  $RT_{NDT}$ , and the reference fracture toughness curve. The thermal stresses resulting from the temperature gradients through the vessel wall are calculated and then the corresponding (thermal) stress intensity factors,  $K_{It}$ , for the reference flaw are computed. From Equation 2, the pressure stress intensity factors are obtained and, from these, the allowable pressures are calculated. For the calculation of the allowable pressure versus coolant temperature during cooldown, the reference flaw of Appendix G to the ASME Code is assumed to exist at the inside of the vessel wall. During cooldown, the controlling location of the flaw is always at the inside of the wall because the thermal gradients produce tensile stresses at the inside, which increase with increasing cooldown rates. Allowable pressure-temperature relations are generated for both steady-state and finite cooldown rate situations. From these relations, composite limit curves are constructed for each cooldown rate of interest.

The use of the composite curve in the cooldown analysis is necessary because control of the cooldown procedure is based on the measurement of reactor coolant temperature, whereas the limiting pressure is actually dependent on the material temperature at the tip of the assumed flaw. During cooldown, the 1/4T vessel location is at a higher temperature than the fluid adjacent to the vessel inner diameter. This condition, of course, is not true for the steady-state situation. It follows that, at any given reactor coolant temperature, the  $\Delta T$  (temperature) developed during cooldown results in a higher value of  $K_{Ic}$  at the 1/4T location for finite cooldown rates than for steady-state operation. Furthermore, if conditions exist so that the increase in  $K_{Ic}$  exceeds  $K_{It}$ , the calculated allowable pressure during cooldown will be greater than the steady-state value. The above procedures are needed because there is no direct control on temperature at the 1/4T location and, therefore, allowable pressures may unknowingly be violated if the rate of cooling is decreased at various intervals along a cooldown ramp. The use of the composite curve eliminates this problem and ensures conservative operation of the system for the entire cooldown period.

Three separate calculations are required to determine the limit curves for finite heatup rates. As is done in the cooldown analysis, allowable pressure-temperature relationships are developed for steady-state conditions as well as finite heatup rate conditions assuming the presence of a 1/4T defect at the inside of the wall. The heatup results in compressive stresses at the inside surface that alleviate the tensile stresses produced by internal pressure. The metal temperature at the crack tip lags the coolant temperature; therefore, the  $K_{Ic}$  for the 1/4T crack during heatup is lower than the  $K_{Ic}$  for the 1/4T crack during steady-state conditions at the same coolant temperature. During heatup, especially at the end of the transient, conditions may exist so that the effects of compressive thermal stresses and lower  $K_{Ic}$  values do not offset each other, and the pressure-temperature curve based on steady-state conditions no longer represents a lower bound of all similar curves for finite heatup rates when the 1/4T flaw is considered. Therefore, both cases have to be analyzed in order to ensure that at any coolant temperature the lower value of the allowable pressure calculated for steady-state and finite heatup rates is obtained.

The second portion of the heatup analysis concerns the calculation of the pressure-temperature limitations for the case in which a 1/4T flaw located at the 1/4T location from the outside surface is assumed. Unlike the situation at the vessel inside surface, the thermal gradients established at the outside surface during heatup produce stresses which are tensile in nature and therefore tend to reinforce any pressure stresses present. These thermal stresses are dependent on both the rate of heatup and the time (or coolant temperature) along the heatup ramp. Since the thermal stresses at the outside are tensile and increase with increasing heatup rates, each heatup rate must be analyzed on an individual basis.

Following the generation of pressure-temperature curves for both the steady-state and finite heatup rate situations, the final limit curves are produced by constructing a composite curve based on a point-by-point comparison of the steady-state and finite heatup rate data. At any given temperature, the allowable pressure is taken to be the lesser of the three values taken from the curves under consideration. The use of the composite curve is necessary to set conservative heatup limitations because it is possible for conditions to exist wherein, over the course of the heatup ramp, the controlling condition switches from the inside to the outside, and the pressure limit must at all times be based on analysis of the most critical criterion.

### 3.3 CLOSURE HEAD/VESSEL FLANGE REQUIREMENTS

10 CFR Part 50, Appendix G [Reference 3] addresses the metal temperature of the closure head flange and vessel flange regions. This rule states that the metal temperature of the closure flange regions must exceed the material unirradiated  $RT_{NDT}$  by at least 120°F for normal operation when the pressure exceeds 20 percent of the preservice hydrostatic test pressure (3107 psi), which is 621 psig for Indian Point Unit 2. The limiting unirradiated  $RT_{NDT}$  of 60°F occurs in both the vessel flange and the closure head flange of the Indian Point Unit 2 reactor vessel, so the minimum allowable temperature of this region is 180°F at pressures greater than 621 psig (without instrument uncertainties). This limit is shown in Figures 5-1 through 5-4 wherever applicable.

#### 4 CALCULATION OF ADJUSTED REFERENCE TEMPERATURE

From Regulatory Guide 1.99, Revision 2, the adjusted reference temperature (ART) for each material in the beltline region is given by the following expression:

$$\text{ART} = \text{Initial RT}_{\text{NDT}} + \Delta\text{RT}_{\text{NDT}} + \text{Margin} \quad (7)$$

Initial  $\text{RT}_{\text{NDT}}$  is the reference temperature for the unirradiated material as defined in paragraph NB-2331 of Section III of the ASME Boiler and Pressure Vessel Code [Reference 12]]. If measured values of initial  $\text{RT}_{\text{NDT}}$  for the material in question are not available, generic mean values for that class of material may be used if there are sufficient test results to establish a mean and standard deviation for the class.

$\Delta\text{RT}_{\text{NDT}}$  is the mean value of the adjustment in reference temperature caused by irradiation and should be calculated as follows:

$$\Delta\text{RT}_{\text{NDT}} = \text{CF} * f^{(0.28 - 0.10 \log f)} \quad (8)$$

To calculate  $\Delta\text{RT}_{\text{NDT}}$  at any depth (e.g., at 1/4T or 3/4T), the following formula must first be used to attenuate the fluence at the specific depth.

$$f_{(\text{depth } x)} = f_{\text{surface}} * e^{(-0.24x)} \quad (9)$$

where  $x$  inches (vessel beltline thickness is 8.625 inches) is the depth into the vessel wall measured from the vessel clad/base metal interface. The resultant fluence is then placed in Equation 8 to calculate the  $\Delta\text{RT}_{\text{NDT}}$  at the specific depth.

Table 4-1 provides neutron exposure information pertinent to the Indian Point Unit 2 (IPP) Stretch Power Uprating Program for the reactor pressure vessel. Neutron fluence information for the Indian Point Unit 2 reactor pressure vessel is based on an assessment of the projected reactor vessel fluence that accounts for a mid-cycle uprate from 3071.4 MWt to 3115 MWt during the current operating Cycle 16 design; incorporation of recently developed loading patterns that are anticipated for use in Cycles 17 through 19 operating at 3216 MWt; and utilization of the Cycle 19 loading pattern with the peripheral assembly powers increased by a factor of 1.05 operating at 3216 MWt from the onset of Cycle 20. In Table 4-1, the calculated maximum fast neutron fluence experienced by the Indian Point Unit 2 reactor pressure vessel is provided as a function of operating time.

The fluence calculations were based on the latest available nuclear cross-section data derived from ENDF/B-VI and made use of the latest available calculational tools for neutron source generation and neutron transport. Furthermore, this neutron transport methodology follows the guidance and meets the requirements of Regulatory Guide 1.190, "Calculational and Dosimetry Methods for Determining Pressure Vessel Neutron Fluence" [Reference 13]. Tables 4-2 and 4-3 provide a summary of the 1/4T and 3/4T fluence and fluence factor values.

Margin is calculated as,  $M = 2 \sqrt{\sigma_i^2 + \sigma_A^2}$ . The standard deviation for the initial  $\text{RT}_{\text{NDT}}$  margin term ( $\sigma_i$ ) is 0°F when the initial  $\text{RT}_{\text{NDT}}$  is a measured value and 17°F when a generic value is

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available. The standard deviation for the  $\Delta RT_{NDT}$  margin term,  $\sigma_{\Delta}$ , is 17°F for plates or forgings, and 8.5°F for plates or forgings when credible surveillance data is used. For welds,  $\sigma_{\Delta}$  is equal to 28°F when surveillance capsule data is not used, and is 14°F (half the value) when credible surveillance capsule data is used.  $\sigma_{\Delta}$  need not exceed 0.5 times the mean value of  $\Delta RT_{NDT}$ . Contained in Tables 4-4 through 4-7 are the Indian Point Unit 2 29.2 and 48 EFPY ART calculations used for generation of the heatup and cooldown curves. Based on a review of the ART values, Intermediate Shell Plate B-2002-3 is the most limiting material in the "Axial-Flaw" case and circumferential weld 34B009 is the limiting material in the "Circ-Flaw" case. Contained in Table 4-8 is a summary of the limiting ART values that will be used in generation of the Indian Point Unit 2 reactor vessel PT limit curves. These limiting curves will be presented in Section 5.

**Table 4-1**  
**Maximum Fast Neutron (E > 1.0 MeV) Fluence Projections at the Pressure Vessel**  
**Clad/Base Metal Interface – Indian Point Unit 2 [Reference 13]**

| Cumulative<br>Operating Time<br>[EFPY] | Neutron Fluence (E > 1.0 MeV) [n/cm <sup>2</sup> ] |           |           |           |
|--|--|-----------|-----------|-----------|
|  | Azimuthal Location                                 |           |           |           |
|  | 0°   | 15°       | 30°       | 45°       |
| 18.7 (EOC 15)                          | 2.775e+18  | 4.472e+18 | 5.391e+18 | 8.074e+18 |
| 20.6 (EOC 16)                          | 2.995e+18  | 4.803e+18 | 5.835e+18 | 8.750e+18 |
| 25.7 (EOC 19)                          | 3.599e+18  | 5.766e+18 | 7.084e+18 | 1.056e+19 |
| 32.0                                   | 4.452e+18  | 7.124e+18 | 8.741e+18 | 1.296e+19 |
| 48.0                                   | 6.619e+18  | 1.057e+19 | 1.295e+19 | 1.906e+19 |

**TABLE 4-2**  
**Summary of the Vessel Surface, 1/4T and 3/4T Fluence Values**  
**used for the Generation of the 29.2 and 48 EFPY Heatup/Cooldown Curves for Indian**  
**Point Unit 2**

| Material  | Surface<br>(n/cm <sup>2</sup> , E > 1.0 MeV) | 1/4T Fluence<br>(n/cm <sup>2</sup> , E > 1.0 MeV) | 3/4T Fluence<br>(n/cm <sup>2</sup> , E > 1.0 MeV) |
|---|--|---|---|
| <b>29.2 EFPY</b>  |  |   |   |
| Intermediate and Lower<br>Shell Axial Welds<br>(Heat # W5214) | 8.005 x 10 <sup>18</sup>                     | 4.771 x 10 <sup>18</sup>                          | 1.695 x 10 <sup>18</sup>                          |
| All other beltline<br>materials                               | 1.189 x 10 <sup>19</sup>                     | 7.087 x 10 <sup>18</sup>                          | 2.517 x 10 <sup>18</sup>                          |
| <b>48 EFPY</b>  |  |   |   |
| Intermediate and Lower<br>Shell Axial Welds<br>(Heat # W5214) | 1.295 x 10 <sup>19</sup>                     | 7.718 x 10 <sup>18</sup>                          | 2.742 x 10 <sup>18</sup>                          |
| All other beltline<br>materials                               | 1.906 x 10 <sup>19</sup>                     | 1.136 x 10 <sup>19</sup>                          | 4.035 x 10 <sup>18</sup>                          |

**TABLE 4-3**  
**Summary of the 1/4T and 3/4T Fluence Factor Values used for the Generation of the**  
**29.2 and 48 EFPY Heatup/Cooldown Curves for Indian Point Unit 2**

| <b>Material</b>   | <b>1/4T Fluence</b><br>(x 10 <sup>19</sup> n/cm <sup>2</sup> , E > 1.0 MeV) | <b>1/4T FF</b> | <b>3/4T Fluence</b><br>(x 10 <sup>19</sup> n/cm <sup>2</sup> , E > 1.0 MeV) | <b>3/4T FF</b> |
|---|---|----------------|---|----------------|
| <b>29.2 EFPY</b>  |   |                |   |                |
| Intermediate and Lower Shell Axial Welds (Heat # W5214) | 0.4771  | 0.7937         | 0.1695  | 0.5306         |
| All other beltline materials                            | 0.7087  | 0.9034         | 0.2517  | 0.6257         |
| <b>48 EFPY</b>  |   |                |   |                |
| Intermediate and Lower Shell Axial Welds (Heat # W5214) | 0.7718  | 0.9273         | 0.2742  | 0.6473         |
| All other beltline materials                            | 1.136   | 1.0356         | 0.4035  | 0.7483         |



**TABLE 4-4**  
**Calculation of the Indian Point Unit 2 ART Values for the 1/4T Location @ 29.2 EFPY**

| Material   | FF     | CF    | $\Delta RT_{NDT}^{(a)}$ | Margin <sup>(b)</sup> | $RT_{NDT(U)}^{(c)}$ | ART <sup>(d)</sup> |
|--|--------|-------|-------------------------|-----------------------|---------------------|--------------------|
|  |        | (°F)  | (°F)                    | (°F)                  | (°F)                | (°F)               |
| Inter. Shell Plate B-2002-1                            | 0.9034 | 144   | 130.09                  | 34.0                  | 34                  | 198                |
| - Using S/C Data                                       | 0.9034 | 114   | 102.99                  | 17.0                  | 34                  | 154                |
| Inter. Shell Plate B-2002-2                            | 0.9034 | 115.1 | 103.98                  | 34.0                  | 21                  | 159                |
| - Using S/C Data                                       | 0.9034 | 118.2 | 106.79                  | 34.0                  | 21                  | 162                |
| Inter. Shell Plate B-2002-3                            | 0.9034 | 176   | 159.00                  | 34.0                  | 21                  | <b>214</b>         |
| - Using S/C Data                                       | 0.9034 | 181.9 | 164.33                  | 17.0                  | 21                  | 202                |
| Lower Shell Plate B-2003-1                             | 0.9034 | 152   | 137.32                  | 34.0                  | 20                  | 191                |
| Lower Shell Plate B-2003-2                             | 0.9034 | 128.8 | 116.36                  | 34.0                  | -20                 | 130                |
| Intermediate & Lower Shell Axial Welds (Heat # W5214)  | 0.7937 | 230.2 | 182.72                  | 65.5                  | -56                 | 192                |
| - Using S/C Data                                       | 0.7937 | 251.8 | 199.9                   | 44.0                  | -56                 | 188                |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009) | 0.9034 | 220.9 | 199.57                  | 65.5                  | -56                 | <b>209</b>         |

**NOTES:**

- (a)  $\Delta RT_{NDT} = CF * FF$ .
- (b) All surveillance data deemed credible with the exception of lower shell plate B-2002-2.
- (c) Initial  $RT_{NDT}$  values are measured values except for the welds.
- (d)  $ART = RT_{NDT(U)} + \Delta RT_{NDT} + Margin$ .

**TABLE 4-5**  
**Calculation of the Indian Point Unit 2 ART Values for the 3/4T Location @ 29.2 EFPY**

| Material   | FF     | CF    | $\Delta RT_{NDT}^{(a)}$ | Margin <sup>(b)</sup> | $RT_{NDT(U)}^{(c)}$ | ART <sup>(d)</sup> |
|--|--------|-------|-------------------------|-----------------------|---------------------|--------------------|
|  |        | (°F)  | (°F)                    | (°F)                  | (°F)                | (°F)               |
| Inter. Shell Plate B-2002-1                            | 0.6257 | 144   | 90.10                   | 34.0                  | 34                  | 158                |
| - Using S/C Data                                       | 0.6257 | 114   | 71.33                   | 17.0                  | 34                  | 122                |
| Inter. Shell Plate B-2002-2                            | 0.6257 | 115.1 | 72.02                   | 34.0                  | 21                  | 127                |
| - Using S/C Data                                       | 0.6257 | 118.2 | 73.96                   | 34.0                  | 21                  | 129                |
| Inter. Shell Plate B-2002-3                            | 0.6257 | 176   | 110.12                  | 34.0                  | 21                  | <b>165</b>         |
| - Using S/C Data                                       | 0.6257 | 181.9 | 113.81                  | 17.0                  | 21                  | 152                |
| Lower Shell Plate B-2003-1                             | 0.6257 | 152   | 95.10                   | 34.0                  | 20                  | 149                |
| Lower Shell Plate B-2003-2                             | 0.6257 | 128.8 | 80.59                   | 34.0                  | -20                 | 95                 |
| Intermediate & Lower Shell Axial Welds (Heat # W5214)  | 0.5306 | 230.2 | 122.14                  | 65.5                  | -56                 | 132                |
| - Using S/C Data                                       | 0.5306 | 251.8 | 133.6                   | 44.0                  | -56                 | 122                |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009) | 0.6257 | 220.9 | 138.21                  | 65.5                  | -56                 | <b>148</b>         |

**NOTES:**

- (a)  $\Delta RT_{NDT} = CF * FF$ .
- (b) All surveillance data deemed credible with the exception of lower shell plate B-2002-2.
- (c) Initial  $RT_{NDT}$  values are measured values except for the welds.
- (d)  $ART = RT_{NDT(U)} + \Delta RT_{NDT} + Margin$ .

TABLE 4-6

## Calculation of the Indian Point Unit 2 ART Values for the 1/4T Location @ 48 EFPY

| Material   | FF     | CF    | $\Delta RT_{NDT}^{(a)}$ | Margin <sup>(b)</sup> | $RT_{NDT(U)}^{(c)}$ | $ART^{(d)}$ |
|--|--------|-------|-------------------------|-----------------------|---------------------|-------------|
|  |        | (°F)  | (°F)                    | (°F)                  | (°F)                | (°F)        |
| Inter. Shell Plate B-2002-1                            | 1.0356 | 144   | 149.13                  | 34.0                  | 34                  | 217         |
| - Using S/C Data                                       | 1.0356 | 114   | 118.06                  | 17.0                  | 34                  | 169         |
| Inter. Shell Plate B-2002-2                            | 1.0356 | 115.1 | 119.20                  | 34.0                  | 21                  | 174         |
| - Using S/C Data                                       | 1.0356 | 118.2 | 122.41                  | 34.0                  | 21                  | 177         |
| Inter. Shell Plate B-2002-3                            | 1.0356 | 176   | 182.27                  | 34.0                  | 21                  | 237         |
| - Using S/C Data                                       | 1.0356 | 181.9 | 188.38                  | 17.0                  | 21                  | 226         |
| Lower Shell Plate B-2003-1                             | 1.0356 | 152   | 157.41                  | 34.0                  | 20                  | 211         |
| Lower Shell Plate B-2003-2                             | 1.0356 | 128.8 | 133.39                  | 34.0                  | -20                 | 147         |
| Intermediate & Lower Shell Axial Welds (Heat # W5214)  | 0.9273 | 230.2 | 213.47                  | 65.5                  | -56                 | 223         |
| - Using S/C Data                                       | 0.9273 | 251.8 | 233.5                   | 44.0                  | -56                 | 222         |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009) | 1.0356 | 220.9 | 228.77                  | 65.5                  | -56                 | 238         |

## NOTES:

- (a)  $\Delta RT_{NDT} = CF * FF$ .  
(b) All surveillance data deemed credible with the exception of lower shell plate B-2002-2.  
(c) Initial  $RT_{NDT}$  values are measured values except for the welds.  
(d)  $ART = RT_{NDT(U)} + \Delta RT_{NDT} + Margin$ .

**TABLE 4-7**  
**Calculation of the Indian Point Unit 2 ART Values for the 3/4T Location @ 48 EFPY**

| Material   | FF     | CF    | $\Delta RT_{NDT}^{(a)}$ | Margin <sup>(b)</sup> | $RT_{NDT(U)}^{(c)}$ | ART <sup>(d)</sup> |
|--|--------|-------|-------------------------|-----------------------|---------------------|--------------------|
|  |        | (°F)  | (°F)                    | (°F)                  | (°F)                | (°F)               |
| Inter. Shell Plate B-2002-1                            | 0.7483 | 144   | 107.76                  | 34.0                  | 34                  | 176                |
| - Using S/C Data                                       | 0.7483 | 114   | 85.31                   | 17.0                  | 34                  | 136                |
| Inter. Shell Plate B-2002-2                            | 0.7483 | 115.1 | 86.13                   | 34.0                  | 21                  | 141                |
| - Using S/C Data                                       | 0.7483 | 118.2 | 88.45                   | 34.0                  | 21                  | 143                |
| Inter. Shell Plate B-2002-3                            | 0.7483 | 176   | 131.71                  | 34.0                  | 21                  | <b>187</b>         |
| - Using S/C Data                                       | 0.7483 | 181.9 | 136.12                  | 17.0                  | 21                  | 174                |
| Lower Shell Plate B-2003-1                             | 0.7483 | 152   | 113.75                  | 34.0                  | 20                  | 168                |
| Lower Shell Plate B-2003-2                             | 0.7483 | 128.8 | 96.39                   | 34.0                  | -20                 | 110                |
| Intermediate & Lower Shell Axial Welds (Heat # W5214)  | 0.6473 | 230.2 | 149.00                  | 65.5                  | -56                 | 159                |
| - Using S/C Data                                       | 0.6473 | 251.8 | 163.0                   | 44.0                  | -56                 | 151                |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009) | 0.7483 | 220.9 | 165.31                  | 65.5                  | -56                 | <b>175</b>         |

**NOTES:**

- (a)  $\Delta RT_{NDT} = CF * FF$ .
- (b) All surveillance data deemed credible with the exception of lower shell plate B-2002-2.
- (c) Initial  $RT_{NDT}$  values are measured values except for the welds.
- (d)  $ART = RT_{NDT(U)} + \Delta RT_{NDT} + Margin$ .

**TABLE 4-8**  
**Summary of the Limiting ART Values Used in the**  
**Generation of the Indian Point Unit 2 Heatup/Cooldown Curves**

| EFPY                | Limiting "Circ-Flaw" ART<br>(Circumferential Weld Heat #34B009) |            | Limiting "Axial-Flaw" ART<br>(Intermediate Shell Plate B-2002-3) |            |
|---------------------|---|------------|--|------------|
|                     | 1/4T (°F)   | 3/4T (°F)  | 1/4T (°F)  | 3/4T (°F)  |
| Indian Point Unit 2 |   |            |  |            |
| 29.2                | <b>209</b>  | <b>148</b> | <b>214</b>   | <b>165</b> |
| 48                  | <b>238</b>  | <b>175</b> | <b>237</b>   | <b>187</b> |

## 5 HEATUP AND COOLDOWN PRESSURE-TEMPERATURE LIMIT CURVES

Pressure-temperature limit curves for normal heatup and cooldown of the primary reactor coolant system have been calculated for the pressure and temperature in the reactor vessel beltline region using the methods discussed in Sections 3 and 4 of this report. This approved methodology is also presented in WCAP-14040-NP-A, Revision 4.

Figure 5-1 presents the limiting heatup curves without margins for possible instrumentation errors using heatup rates of 60 and 100°F/hr applicable for 29.2 EFPY with the "Flange-Notch" requirement using the "Axial-flaw" methodology. This curve was generated using the 1998 ASME Code Section XI, Appendix G. Figure 5-2 presents the limiting cooldown curve without margins for possible instrumentation errors using cooldown rates of 0, 20, 40, 60 and 100°F/hr applicable for 29.2 EFPY with the "Flange-Notch" requirement. Again, this curve was generated using the 1998 ASME Code Section XI, Appendix G. These PT limit curves bound those generated using the "Circ-flaw" methodology with the limiting circ-weld ART value from circumferential weld B34009.

Figure 5-3 presents the limiting heatup curves without margins for possible instrumentation errors using heatup rates of 60 and 100°F/hr applicable for 48 EFPY with the "Flange-Notch" requirement using the "Axial-flaw" methodology. This curve was generated using the 1998 ASME Code Section XI, Appendix G. Figure 5-4 presents the limiting cooldown curve without margins for possible instrumentation errors using cooldown rates of 0, 20, 40, 60 and 100°F/hr applicable for 48 EFPY with the "Flange-Notch" requirement. Again, this curve was generated using the 1998 ASME Code Section XI, Appendix G. These PT limit curves bound those generated using the "Circ-flaw" methodology with the limiting circ-weld ART value from circumferential weld B34009.

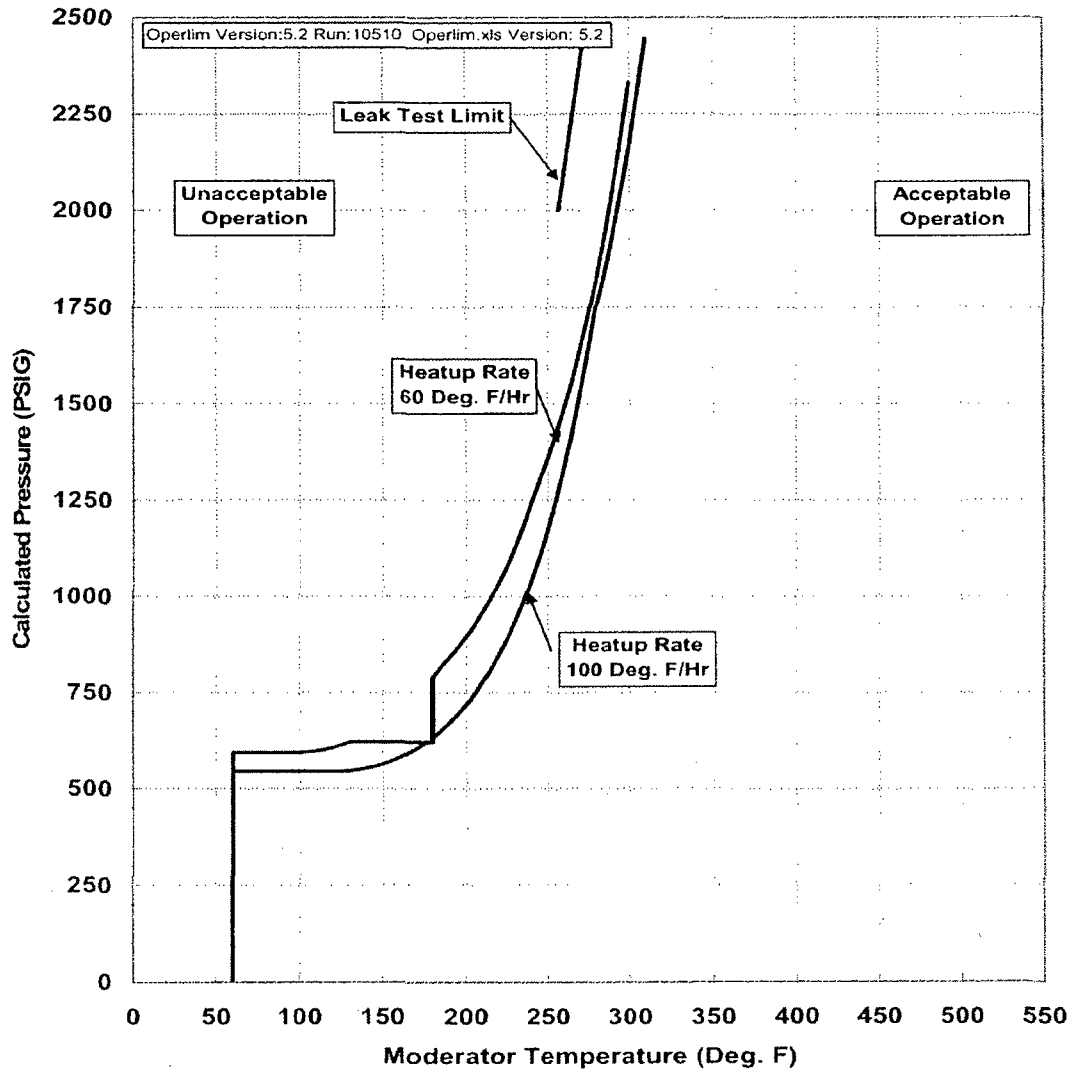
Allowable combinations of temperature and pressure for specific temperature change rates are below and to the right of the limit line shown in Figures 5-1 through 5-4. The criticality limit curve that specifies pressure-temperature limits for core operation have not been included in this report per customer request.

Figures 5-1 through 5-4 define all of the above limits for ensuring prevention of non-ductile failure for the Indian Point Unit 2 reactor vessel for 29.2 and 48 EFPY with the "Flange-Notch" requirement (without instrumentation uncertainties). The data points used for developing the heatup and cooldown pressure-temperature limit curves shown in Figures 5-1 through 5-4 are presented in Tables 5-1 through 5-4. The resulting data for the "Circ-flaw" methodology is presented in Appendix B.

MATERIAL PROPERTY BASIS

LIMITING MATERIAL: Intermediate Shell Plate B-2002-3

LIMITING ART VALUES AT 29.2 EFPY:     1/4T,  214°F  
   3/4T,  165°F



**Figure 5-1** Indian Point Unit 2 Reactor Coolant System Heatup Limitations (Heatup Rates of 60 and 100°F/hr) Applicable for 29.2 EFPY (w/ the “Flange-Notch” and w/o Margins for Instrumentation Errors) Using 1998 App. G Methodology (w/K<sub>IC</sub>)

MATERIAL PROPERTY BASIS

LIMITING MATERIAL: Intermediate Shell Plate B-2002-3  
 LIMITING ART VALUES AT 29.2 EFPY: 1/4T, 214°F

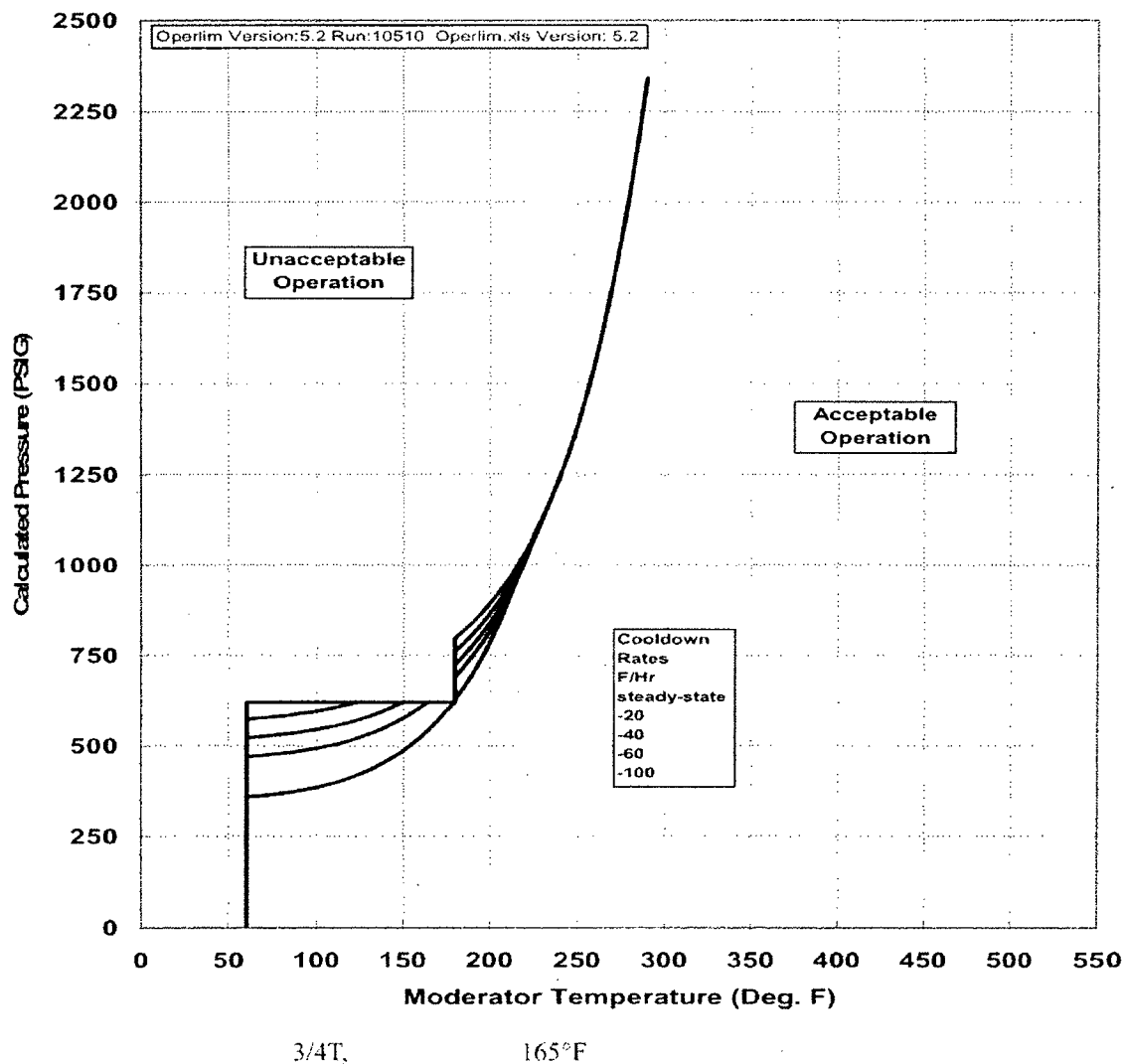


Figure 5-2 Indian Point Unit 2 Reactor Coolant System Cooldown Limitations (Cooldown Rates up to 100°F/hr) Applicable for 29.2 EFPY (w/ the “Flange-Notch” and w/o Margins for Instrumentation Errors) Using 1998 App. G Methodology (w/K<sub>1c</sub>)



MATERIAL PROPERTY BASIS

LIMITING MATERIAL: Intermediate Shell Plate B-2002-3

LIMITING ART VALUES AT 48 EFY: 1/4T, 237°F

3/4T, 187°F

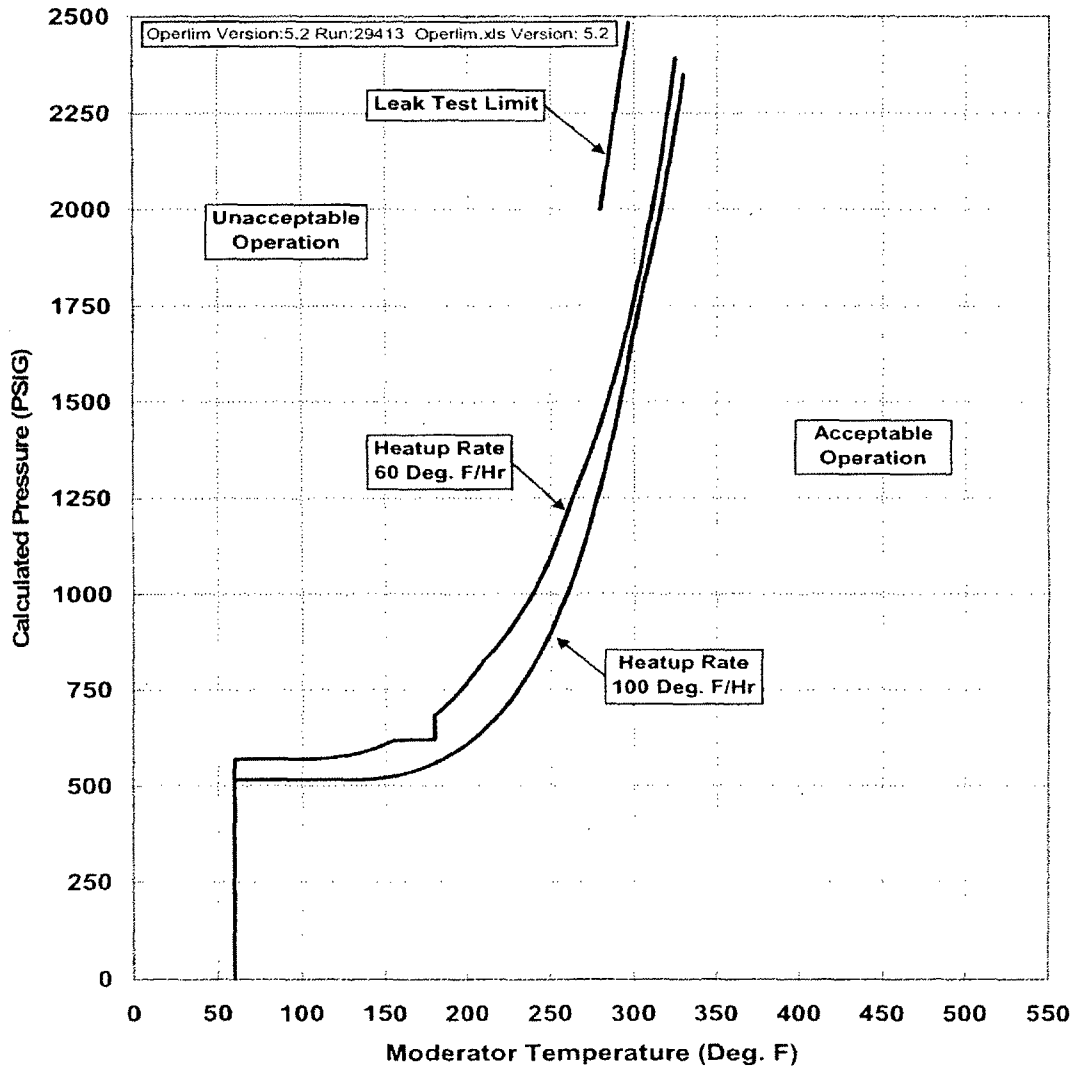


Figure 5-3 Indian Point Unit 2 Reactor Coolant System Heatup Limitations (Heatup Rates of 60 and 100°F/hr) Applicable for 48 EFY (w/ the “Flange-Notch” and w/o Margins for Instrumentation Errors) Using 1998 App. G Methodology (w/ $K_{1c}$ )

MATERIAL PROPERTY BASIS

LIMITING MATERIAL: Intermediate Shell Plate B-2002-3

LIMITING ART VALUES AT 48 EFPY:     1/4T, 237°F  
  3/4T, 187°F

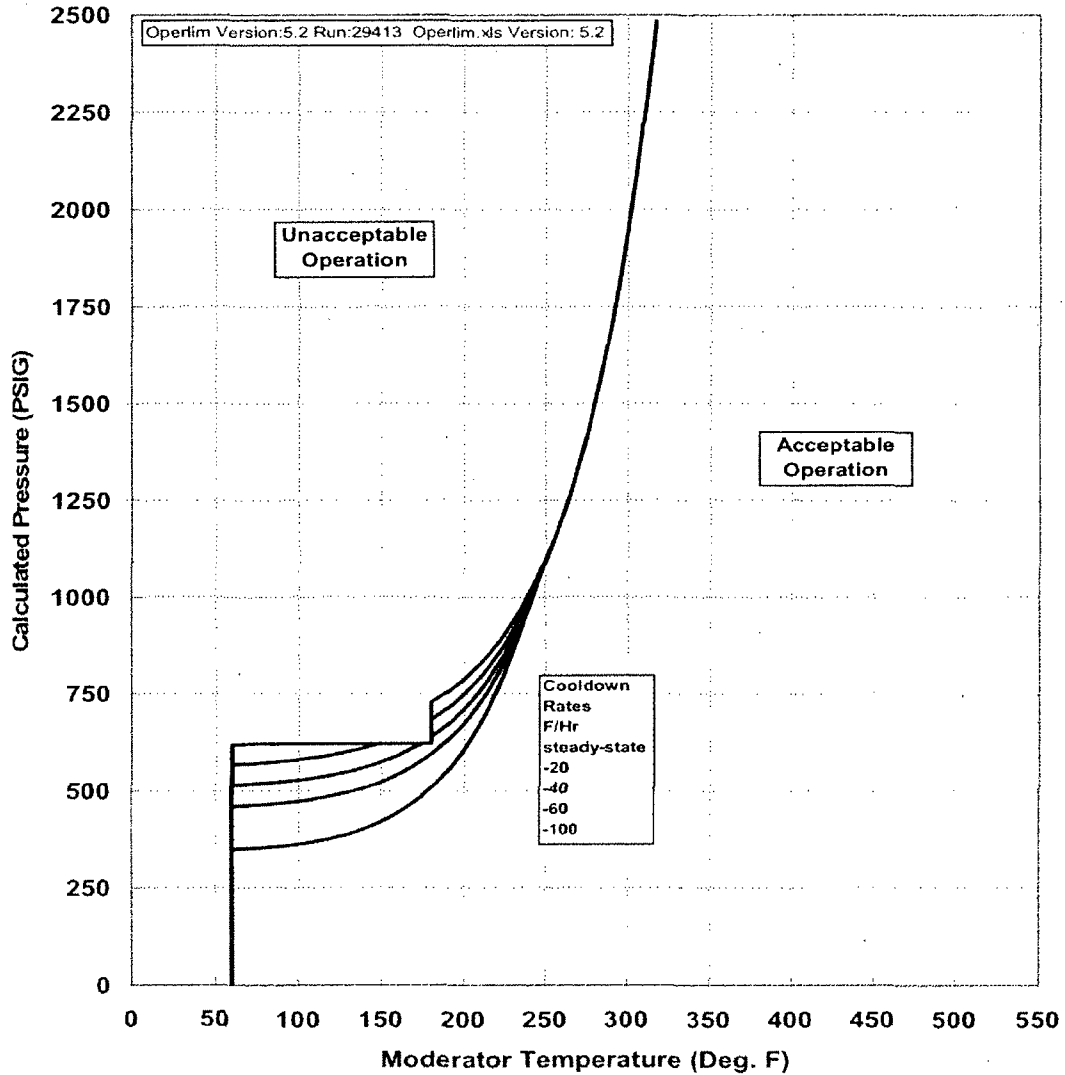


Figure 5-4 Indian Point Unit 2 Reactor Coolant System Cooldown Limitations (Cooldown Rates up to 100°F/hr) Applicable for 48 EFPY (w/ the “Flange-Notch” and w/o Margins for Instrumentation Errors) Using 1998 App. G Methodology (w/K<sub>1c</sub>)

TABLE 5-1

**29.2 EFPY Heatup Curve Data Points Using 1998 App. G Methodology  
(w/ $K_{IC}$ , w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| 60°F/hr      |                 | 100°F/hr     |                 |
|--------------|-----------------|--------------|-----------------|
| Temp<br>(°F) | Press<br>(psig) | Temp<br>(°F) | Press<br>(psig) |
| 60           | 0               | 60           | 0               |
| 60           | 594             | 60           | 546             |
| 65           | 594             | 65           | 546             |
| 70           | 594             | 70           | 546             |
| 75           | 594             | 75           | 546             |
| 80           | 594             | 80           | 546             |
| 85           | 594             | 85           | 546             |
| 90           | 594             | 90           | 546             |
| 95           | 594             | 95           | 546             |
| 100          | 594             | 100          | 546             |
| 105          | 596             | 105          | 546             |
| 110          | 598             | 110          | 546             |
| 115          | 602             | 115          | 546             |
| 120          | 607             | 120          | 546             |
| 125          | 614             | 125          | 546             |
| 130          | 621             | 130          | 547             |
| 135          | 621             | 135          | 550             |
| 140          | 621             | 140          | 553             |
| 145          | 621             | 145          | 558             |
| 150          | 621             | 150          | 564             |
| 155          | 621             | 155          | 572             |
| 160          | 621             | 160          | 581             |
| 165          | 621             | 165          | 592             |
| 170          | 621             | 170          | 604             |
| 175          | 621             | 175          | 618             |
| 180          | 621             | 180          | 621             |
| 180          | 789             | 180          | 633             |
| 185          | 818             | 185          | 651             |
| 190          | 842             | 190          | 671             |
| 195          | 866             | 195          | 693             |
| 200          | 894             | 200          | 718             |
| 205          | 924             | 205          | 745             |
| 210          | 957             | 210          | 776             |
| 215          | 994             | 215          | 810             |
| 220          | 1034            | 220          | 847             |
| 225          | 1079            | 225          | 889             |
| 230          | 1129            | 230          | 935             |
| 235          | 1184            | 235          | 986             |
| 240          | 1245            | 240          | 1042            |
| 245          | 1302            | 245          | 1104            |
| 250          | 1357            | 250          | 1173            |
| 255          | 1417            | 255          | 1249            |
| 260          | 1484            | 260          | 1332            |
| 265          | 1557            | 265          | 1425            |
| 270          | 1638            | 270          | 1526            |
| 275          | 1728            | 275          | 1639            |

| 60°F/hr                |              | 100°F/hr  |              |
|------------------------|--------------|-----------|--------------|
| Temp (°F)              | Press (psig) | Temp (°F) | Press (psig) |
| 280                    | 1826         | 280       | 1753         |
| 285                    | 1935         | 285       | 1842         |
| 290                    | 2055         | 290       | 1941         |
| 295                    | 2188         | 295       | 2049         |
| 300                    | 2334         | 300       | 2169         |
| 304.7                  | 2485         | 305       | 2300         |
|                        |              | 310       | 2446         |
|                        |              | 311.2     | 2485         |
|                        |              |           |              |
| <b>Leak Test Limit</b> |              | 257       | 2000         |
|                        |              | 274       | 2485         |

**TABLE 5-2**  
**29.2 EFPY Cooldown Curve Data Points Using 1998 App. G Methodology**  
**(w/K<sub>IC</sub>, w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 60           | 0            | 60        | 0            | 60        | 0            | 60        | 0            | 60        | 0            |
| 60           | 621          | 60        | 574          | 60        | 522          | 60        | 470          | 60        | 360          |
| 65           | 621          | 65        | 576          | 65        | 524          | 65        | 471          | 65        | 362          |
| 70           | 621          | 70        | 578          | 70        | 526          | 70        | 473          | 70        | 364          |
| 75           | 621          | 75        | 580          | 75        | 528          | 75        | 476          | 75        | 366          |
| 80           | 621          | 80        | 583          | 80        | 531          | 80        | 478          | 80        | 369          |
| 85           | 621          | 85        | 585          | 85        | 534          | 85        | 481          | 85        | 372          |
| 90           | 621          | 90        | 588          | 90        | 537          | 90        | 485          | 90        | 376          |
| 95           | 621          | 95        | 592          | 95        | 541          | 95        | 488          | 95        | 381          |
| 100          | 621          | 100       | 596          | 100       | 545          | 100       | 493          | 100       | 386          |
| 105          | 621          | 105       | 600          | 105       | 549          | 105       | 497          | 105       | 391          |
| 110          | 621          | 110       | 605          | 110       | 554          | 110       | 503          | 110       | 397          |
| 115          | 621          | 115       | 610          | 115       | 560          | 115       | 509          | 115       | 405          |
| 120          | 621          | 120       | 616          | 120       | 566          | 120       | 516          | 120       | 413          |
| 125          | 621          | 125       | 621          | 125       | 573          | 125       | 523          | 125       | 422          |
| 130          | 621          | 130       | 621          | 130       | 581          | 130       | 532          | 130       | 432          |
| 135          | 621          | 135       | 621          | 135       | 590          | 135       | 541          | 135       | 443          |
| 140          | 621          | 140       | 621          | 140       | 599          | 140       | 552          | 140       | 456          |
| 145          | 621          | 145       | 621          | 145       | 610          | 145       | 563          | 145       | 470          |
| 150          | 621          | 150       | 621          | 150       | 621          | 150       | 576          | 150       | 486          |
| 155          | 621          | 155       | 621          | 155       | 621          | 155       | 591          | 155       | 504          |
| 160          | 621          | 160       | 621          | 160       | 621          | 160       | 607          | 160       | 524          |
| 165          | 621          | 165       | 621          | 165       | 621          | 165       | 621          | 165       | 546          |
| 170          | 621          | 170       | 621          | 170       | 621          | 170       | 621          | 170       | 570          |
| 175          | 621          | 175       | 621          | 175       | 621          | 175       | 621          | 175       | 597          |
| 180          | 621          | 180       | 621          | 180       | 621          | 180       | 621          | 180       | 621          |
| 180          | 799          | 180       | 762          | 180       | 726          | 180       | 691          | 180       | 628          |
| 185          | 819          | 185       | 784          | 185       | 750          | 185       | 718          | 185       | 661          |
| 190          | 842          | 190       | 809          | 190       | 777          | 190       | 748          | 190       | 699          |
| 195          | 866          | 195       | 836          | 195       | 807          | 195       | 782          | 195       | 740          |
| 200          | 894          | 200       | 866          | 200       | 841          | 200       | 819          | 200       | 787          |
| 205          | 924          | 205       | 899          | 205       | 878          | 205       | 860          | 205       | 838          |
| 210          | 957          | 210       | 936          | 210       | 918          | 210       | 905          | 210       | 895          |
| 215          | 994          | 215       | 977          | 215       | 964          | 215       | 955          | 215       | 955          |
| 220          | 1034         | 220       | 1022         | 220       | 1014         | 220       | 1011         | 220       | 1011         |
| 225          | 1079         | 225       | 1072         | 225       | 1069         | 225       | 1069         | 225       | 1069         |
| 230          | 1129         | 230       | 1127         | 230       | 1127         | 230       | 1127         | 230       | 1127         |
| 235          | 1184         | 235       | 1184         | 235       | 1184         | 235       | 1184         | 235       | 1184         |
| 240          | 1245         | 240       | 1245         | 240       | 1245         | 240       | 1245         | 240       | 1245         |
| 245          | 1312         | 245       | 1312         | 245       | 1312         | 245       | 1312         | 245       | 1312         |
| 250          | 1386         | 250       | 1386         | 250       | 1386         | 250       | 1386         | 250       | 1386         |
| 255          | 1468         | 255       | 1468         | 255       | 1468         | 255       | 1468         | 255       | 1468         |
| 260          | 1558         | 260       | 1558         | 260       | 1558         | 260       | 1558         | 260       | 1558         |
| 265          | 1658         | 265       | 1658         | 265       | 1658         | 265       | 1658         | 265       | 1658         |
| 270          | 1769         | 270       | 1769         | 270       | 1769         | 270       | 1769         | 270       | 1769         |
| 275          | 1891         | 275       | 1891         | 275       | 1891         | 275       | 1891         | 275       | 1891         |

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 280          | 2026         | 280       | 2026         | 280       | 2026         | 280       | 2026         | 280       | 2026         |
| 285          | 2176         | 285       | 2176         | 285       | 2176         | 285       | 2176         | 285       | 2176         |
| 290          | 2341         | 290       | 2341         | 290       | 2341         | 290       | 2341         | 290       | 2341         |
| 294          | 2485         | 294       | 2485         | 294       | 2485         | 294       | 2485         | 294       | 2485         |

**TABLE 5-3**  
**48 EFPY Heatup Curve Data Points Using 1998 App. G Methodology**  
**(w/K<sub>IC</sub>, w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| 60°F/hr   |              | 100°F/hr  |              |
|-----------|--------------|-----------|--------------|
| Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 60        | 0            | 60        | 0            |
| 60        | 570          | 60        | 516          |
| 65        | 570          | 65        | 516          |
| 70        | 570          | 70        | 516          |
| 75        | 570          | 75        | 516          |
| 80        | 570          | 80        | 516          |
| 85        | 570          | 85        | 516          |
| 90        | 570          | 90        | 516          |
| 95        | 570          | 95        | 516          |
| 100       | 570          | 100       | 516          |
| 105       | 570          | 105       | 516          |
| 110       | 571          | 110       | 516          |
| 115       | 572          | 115       | 516          |
| 120       | 574          | 120       | 516          |
| 125       | 578          | 125       | 516          |
| 130       | 582          | 130       | 516          |
| 135       | 587          | 135       | 516          |
| 140       | 594          | 140       | 517          |
| 145       | 601          | 145       | 519          |
| 150       | 609          | 150       | 522          |
| 155       | 619          | 155       | 525          |
| 160       | 621          | 160       | 530          |
| 165       | 621          | 165       | 536          |
| 170       | 621          | 170       | 543          |
| 175       | 621          | 175       | 551          |
| 180       | 621          | 180       | 560          |
| 180       | 686          | 185       | 571          |
| 185       | 704          | 190       | 583          |
| 190       | 725          | 195       | 596          |
| 195       | 747          | 200       | 612          |
| 200       | 772          | 205       | 629          |
| 205       | 800          | 210       | 648          |
| 210       | 828          | 215       | 669          |
| 215       | 851          | 220       | 693          |
| 220       | 877          | 225       | 719          |
| 225       | 905          | 230       | 749          |
| 230       | 937          | 235       | 781          |
| 235       | 971          | 240       | 817          |
| 240       | 1010         | 245       | 856          |
| 245       | 1052         | 250       | 900          |
| 250       | 1099         | 255       | 949          |
| 255       | 1150         | 260       | 1002         |
| 260       | 1208         | 265       | 1061         |
| 265       | 1271         | 270       | 1126         |
| 270       | 1324         | 275       | 1198         |
| 275       | 1381         | 280       | 1278         |

| 60°F/hr                |              | 100°F/hr  |              |
|------------------------|--------------|-----------|--------------|
| Temp (°F)              | Press (psig) | Temp (°F) | Press (psig) |
| 280                    | 1443         | 285       | 1365         |
| 285                    | 1512         | 290       | 1461         |
| 290                    | 1588         | 295       | 1568         |
| 295                    | 1672         | 300       | 1685         |
| 300                    | 1765         | 305       | 1786         |
| 305                    | 1867         | 310       | 1878         |
| 310                    | 1980         | 315       | 1980         |
| 315                    | 2104         | 320       | 2092         |
| 320                    | 2241         | 325       | 2215         |
| 325                    | 2393         | 330       | 2351         |
| 327.8                  | 2485         | 334.5     | 2485         |
|                        |              | 280       | 297          |
| <b>Leak Test Limit</b> |              | 2000      | 2485         |



**TABLE 5-4**  
**48 EFPY Cooldown Curve Data Points Using 1998 App. G Methodology**  
**(w/K<sub>IC</sub>, w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 60           | 0            | 60        | 0            | 60        | 0            | 60        | 0            | 60        | 0            |
| 60           | 618          | 60        | 567          | 60        | 515          | 60        | 461          | 60        | 349          |
| 65           | 619          | 65        | 568          | 65        | 516          | 65        | 462          | 65        | 350          |
| 70           | 620          | 70        | 569          | 70        | 517          | 70        | 463          | 70        | 351          |
| 75           | 621          | 75        | 571          | 75        | 518          | 75        | 464          | 75        | 352          |
| 80           | 621          | 80        | 572          | 80        | 519          | 80        | 465          | 80        | 353          |
| 85           | 621          | 85        | 574          | 85        | 521          | 85        | 467          | 85        | 355          |
| 90           | 621          | 90        | 576          | 90        | 523          | 90        | 469          | 90        | 357          |
| 95           | 621          | 95        | 578          | 95        | 525          | 95        | 471          | 95        | 359          |
| 100          | 621          | 100       | 580          | 100       | 527          | 100       | 473          | 100       | 362          |
| 105          | 621          | 105       | 583          | 105       | 530          | 105       | 476          | 105       | 365          |
| 110          | 621          | 110       | 585          | 110       | 533          | 110       | 479          | 110       | 368          |
| 115          | 621          | 115       | 589          | 115       | 536          | 115       | 483          | 115       | 373          |
| 120          | 621          | 120       | 592          | 120       | 540          | 120       | 487          | 120       | 377          |
| 125          | 621          | 125       | 596          | 125       | 544          | 125       | 491          | 125       | 382          |
| 130          | 621          | 130       | 601          | 130       | 549          | 130       | 496          | 130       | 388          |
| 135          | 621          | 135       | 606          | 135       | 554          | 135       | 502          | 135       | 395          |
| 140          | 621          | 140       | 611          | 140       | 560          | 140       | 508          | 140       | 403          |
| 145          | 621          | 145       | 617          | 145       | 567          | 145       | 516          | 145       | 411          |
| 150          | 621          | 150       | 621          | 150       | 574          | 150       | 523          | 150       | 421          |
| 155          | 621          | 155       | 621          | 155       | 582          | 155       | 532          | 155       | 431          |
| 160          | 621          | 160       | 621          | 160       | 591          | 160       | 542          | 160       | 443          |
| 165          | 621          | 165       | 621          | 165       | 601          | 165       | 553          | 165       | 457          |
| 170          | 621          | 170       | 621          | 170       | 613          | 170       | 565          | 170       | 472          |
| 175          | 621          | 175       | 621          | 175       | 621          | 175       | 579          | 175       | 489          |
| 180          | 621          | 180       | 621          | 180       | 621          | 180       | 594          | 180       | 507          |
| 180          | 728          | 180       | 683          | 180       | 639          | 185       | 611          | 185       | 528          |
| 185          | 741          | 185       | 697          | 185       | 654          | 190       | 630          | 190       | 551          |
| 190          | 755          | 190       | 713          | 190       | 671          | 195       | 651          | 195       | 577          |
| 195          | 771          | 195       | 730          | 195       | 690          | 200       | 674          | 200       | 606          |
| 200          | 788          | 200       | 749          | 200       | 711          | 205       | 699          | 205       | 638          |
| 205          | 807          | 205       | 770          | 205       | 734          | 210       | 728          | 210       | 673          |
| 210          | 828          | 210       | 793          | 210       | 759          | 215       | 759          | 215       | 712          |
| 215          | 851          | 215       | 818          | 215       | 788          | 220       | 794          | 220       | 756          |
| 220          | 877          | 220       | 847          | 220       | 819          | 225       | 833          | 225       | 805          |
| 225          | 905          | 225       | 878          | 225       | 854          | 230       | 876          | 230       | 859          |
| 230          | 937          | 230       | 913          | 230       | 892          | 235       | 924          | 235       | 918          |
| 235          | 971          | 235       | 951          | 235       | 935          | 240       | 976          | 240       | 976          |
| 240          | 1010         | 240       | 994          | 240       | 982          | 245       | 1035         | 245       | 1035         |
| 245          | 1052         | 245       | 1041         | 245       | 1035         | 250       | 1093         | 250       | 1093         |
| 250          | 1099         | 250       | 1093         | 250       | 1093         | 255       | 1150         | 255       | 1150         |
| 255          | 1150         | 255       | 1150         | 255       | 1150         | 260       | 1208         | 260       | 1208         |
| 260          | 1208         | 260       | 1208         | 260       | 1208         | 265       | 1271         | 265       | 1271         |
| 265          | 1271         | 265       | 1271         | 265       | 1271         | 270       | 1341         | 270       | 1341         |
| 270          | 1341         | 270       | 1341         | 270       | 1341         | 275       | 1418         | 275       | 1418         |
| 275          | 1418         | 275       | 1418         | 275       | 1418         | 280       | 1503         | 280       | 1503         |

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 280          | 1503         | 280       | 1503         | 280       | 1503         | 285       | 1597         | 285       | 1597         |
| 285          | 1597         | 285       | 1597         | 285       | 1597         | 290       | 1701         | 290       | 1701         |
| 290          | 1701         | 290       | 1701         | 290       | 1701         | 295       | 1816         | 295       | 1816         |
| 295          | 1816         | 295       | 1816         | 295       | 1816         | 300       | 1944         | 300       | 1944         |
| 300          | 1944         | 300       | 1944         | 300       | 1944         | 305       | 2084         | 305       | 2084         |
| 305          | 2084         | 305       | 2084         | 305       | 2084         | 310       | 2240         | 310       | 2240         |
| 310          | 2240         | 310       | 2240         | 310       | 2240         | 315       | 2411         | 315       | 2411         |
| 315          | 2411         | 315       | 2411         | 315       | 2411         | 317       | 2485         | 317       | 2485         |
| 317          | 2485         | 317       | 2485         | 317       | 2485         |           |              |           |              |

## 6 REFERENCES

1. Regulatory Guide 1.99, Revision 2, "Radiation Embrittlement of Reactor Vessel Materials," U.S. Nuclear Regulatory Commission, May 1988.
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3. Code of Federal Regulations, 10 CFR Part 50, Appendix G, "Fracture Toughness Requirements," U.S. Nuclear Regulatory Commission, Washington, D.C., Federal Register, Volume 60, No. 243, dated December 19, 1995.
4. "Fracture Toughness Requirements", Branch Technical Position MTEB 5-2, Chapter 5.3.2 in Standard Review Plan for the Review of Safety Analysis Reports for Nuclear Power Plants, LWR Edition, NUREG-0800, 1981.
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10. CPL-96-203, "Robinson Unit 2 Surveillance Capsule Charpy Test Results", P. A. Grendys, March 6, 1996.
11. Section XI of the ASME Boiler and Pressure Vessel Code, Appendix G, "Fracture Toughness Criteria for Protection Against Failure." Dated December 1998, through 2000 Addendum.
12. 1989 Section III, Division 1 of the ASME Boiler and Pressure Vessel Code, Paragraph NB-2331, Test Requirements and Acceptance Standards, "Material for Vessels."
13. Regulatory Guide 1.190, "Calculational and Dosimetry Methods for Determining Pressure Vessel Neutron Fluence," March 2001.
14. LTR-REA-03-89, "Neutron Exposure Projections to Support the Indian Point Unit 2 Stretch Power Up-rating Program," S. L. Anderson and G. K. Roberts, June 18, 2003.
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**APPENDIX A**  
**Thermal Stress Intensity Factors ( $K_{I_t}$ )**

The following pages contain the thermal stress intensity factors ( $K_{I_t}$ ) for the maximum heatup and cooldown rates. The vessel radii to the  $1/4T$  and  $3/4T$  locations are as follows:

- $1/4T$  Radius = 88.875"
- $3/4T$  Radius = 93.188"

TABLE A-1  
 $K_{It}$  Values for 100°F/hr Heatup Curve for Indian Point Unit 2

| Water Temp. (°F) | Vessel Temperature @ 1/4T Location for 100°F/hr Heatup (°F) | 1/4T Thermal Stress Intensity Factor (KSI SQ. RT. IN.) | Vessel Temperature @ 3/4T Location for 100°F/hr Heatup (°F) | 3/4T Thermal Stress Intensity Factor (KSI SQ. RT. IN.) |
|------------------|---|--|---|--|
| 60               | 56  | -0.995   | 55  | 0.473  |
| 65               | 59  | -2.452   | 55  | 1.438  |
| 70               | 62  | -3.712   | 56  | 2.426  |
| 75               | 65  | -4.910   | 57  | 3.356  |
| 80               | 68  | -5.945   | 59  | 4.190  |
| 85               | 72  | -6.892   | 61  | 4.938  |
| 90               | 76  | -7.714   | 63  | 5.599  |
| 95               | 80  | -8.465   | 65  | 6.192  |
| 100              | 84  | -9.123   | 68  | 6.719  |
| 105              | 88  | -9.721   | 71  | 7.189  |
| 110              | 92  | -10.247  | 75  | 7.609  |
| 115              | 97  | -10.728  | 78  | 7.988  |
| 120              | 101   | -11.154  | 82  | 8.327  |
| 125              | 106   | -11.544  | 85  | 8.634  |
| 130              | 110   | -11.891  | 89  | 8.910  |
| 135              | 115   | -12.210  | 93  | 9.160  |
| 140              | 119   | -12.495  | 97  | 9.387  |
| 145              | 124   | -12.759  | 102   | 9.593  |
| 150              | 129   | -12.996  | 106   | 9.781  |
| 155              | 133   | -13.217  | 110   | 9.953  |
| 160              | 138   | -13.417  | 115   | 10.111   |
| 165              | 143   | -13.604  | 119   | 10.257   |
| 170              | 147   | -13.774  | 123   | 10.391   |
| 175              | 152   | -13.934  | 128   | 10.515   |
| 180              | 157   | -14.082  | 133   | 10.631   |
| 185              | 162   | -14.221  | 137   | 10.739   |
| 190              | 167   | -14.351  | 142   | 10.840   |
| 195              | 172   | -14.474  | 146   | 10.936   |
| 200              | 176   | -14.589  | 151   | 11.026   |

**TABLE A-2**  
**K<sub>II</sub> Values for 100°F/hr Cooldown Curve for Indian Point Unit 2**

| <b>Water Temp. (°F)</b> | <b>Vessel Temperature @ 1/4T Location for 100°F/hr Cooldown (°F)</b> | <b>100°F/hr Cooldown 1/4T Thermal Stress Intensity Factor (KSI SQ. RT. IN.)</b> |
|-------------------------|--|---|
| 200.000                 | 227  | 17.002  |
| 195.000                 | 222  | 16.933  |
| 190.000                 | 217  | 16.864  |
| 185.000                 | 212  | 16.795  |
| 180.000                 | 207  | 16.726  |
| 175.000                 | 201  | 16.656  |
| 170.000                 | 196  | 16.587  |
| 165.000                 | 191  | 16.517  |
| 160.000                 | 186  | 16.448  |
| 155.000                 | 181  | 16.378  |
| 150.000                 | 176  | 16.309  |
| 145.000                 | 171  | 16.239  |
| 140.000                 | 166  | 16.170  |
| 135.000                 | 161  | 16.100  |
| 130.000                 | 156  | 16.031  |
| 125.000                 | 151  | 15.962  |
| 120.000                 | 145  | 15.893  |
| 115.000                 | 140  | 15.824  |
| 110.000                 | 135  | 15.755  |
| 105.000                 | 130  | 15.686  |
| 100.000                 | 125  | 15.617  |
| 95.000                  | 120  | 15.548  |
| 90.000                  | 115  | 15.480  |
| 85.000                  | 110  | 15.412  |
| 80.000                  | 105  | 15.344  |
| 75.000                  | 100  | 15.275  |
| 70.000                  | 95   | 15.207  |
| 65.000                  | 90   | 15.139  |
| 60.000                  | 84   | 15.071  |

**APPENDIX B**  
**Indian Point Unit 2 PT Limit Curve "Circ-Flaw" Methodology Data Results**  
**for 29.2 and 48 EFPY**



TABLE B-1

29.2 EFPY Heatup Curve Data Points Using 1998 App. G "Circ-Flaw" Methodology  
(w/K<sub>IC</sub>, w/Flange Notch and w/o Uncertainties for Instrumentation Errors)

| 60°F/hr                |              | 100°F/hr  |              |
|------------------------|--------------|-----------|--------------|
| Temp (°F)              | Press (psig) | Temp (°F) | Press (psig) |
| 60                     | 0            | 60        | 0            |
| 60                     | 621          | 60        | 621          |
| 65                     | 621          | 65        | 621          |
| 70                     | 621          | 70        | 621          |
| 75                     | 621          | 75        | 621          |
| 80                     | 621          | 80        | 621          |
| 85                     | 621          | 85        | 621          |
| 90                     | 621          | 90        | 621          |
| 95                     | 621          | 95        | 621          |
| 100                    | 621          | 100       | 621          |
| 105                    | 621          | 105       | 621          |
| 110                    | 621          | 110       | 621          |
| 115                    | 621          | 115       | 621          |
| 120                    | 621          | 120       | 621          |
| 125                    | 621          | 125       | 621          |
| 130                    | 621          | 130       | 621          |
| 135                    | 621          | 135       | 621          |
| 140                    | 621          | 140       | 621          |
| 145                    | 621          | 145       | 621          |
| 150                    | 621          | 150       | 621          |
| 155                    | 621          | 155       | 621          |
| 160                    | 621          | 160       | 621          |
| 165                    | 621          | 165       | 621          |
| 170                    | 621          | 170       | 621          |
| 175                    | 621          | 175       | 621          |
| 180                    | 621          | 180       | 621          |
| 180                    | 1713         | 180       | 1444         |
| 185                    | 1759         | 185       | 1496         |
| 190                    | 1811         | 190       | 1553         |
| 195                    | 1868         | 195       | 1618         |
| 200                    | 1931         | 200       | 1689         |
| 205                    | 2000         | 205       | 1768         |
| 210                    | 2077         | 210       | 1856         |
| 215                    | 2162         | 215       | 1953         |
| 220                    | 2256         | 220       | 2061         |
| 225                    | 2360         | 225       | 2180         |
| 230                    | 2475         | 230       | 2312         |
|                        |              | 235       | 2457         |
|                        |              |           |              |
| <b>Leak Test Limit</b> |              | 147       | 195          |
|                        |              | 2000      | 2485         |

**TABLE B-2**  
**29.2 EFPY Cooldown Curve Data Points Using 1998 App. G "Circ-Flaw" Methodology**  
**(w/ $K_{IC}$ , w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 60           | 0            | 60        | 0            | 60        | 0            | 60        | 0            | 60        | 0            |
| 60           | 621          | 60        | 621          | 60        | 621          | 60        | 621          | 60        | 621          |
| 65           | 621          | 65        | 621          | 65        | 621          | 65        | 621          | 65        | 621          |
| 70           | 621          | 70        | 621          | 70        | 621          | 70        | 621          | 70        | 621          |
| 75           | 621          | 75        | 621          | 75        | 621          | 75        | 621          | 75        | 621          |
| 80           | 621          | 80        | 621          | 80        | 621          | 80        | 621          | 80        | 621          |
| 85           | 621          | 85        | 621          | 85        | 621          | 85        | 621          | 85        | 621          |
| 90           | 621          | 90        | 621          | 90        | 621          | 90        | 621          | 90        | 621          |
| 95           | 621          | 95        | 621          | 95        | 621          | 95        | 621          | 95        | 621          |
| 100          | 621          | 100       | 621          | 100       | 621          | 100       | 621          | 100       | 621          |
| 105          | 621          | 105       | 621          | 105       | 621          | 105       | 621          | 105       | 621          |
| 110          | 621          | 110       | 621          | 110       | 621          | 110       | 621          | 110       | 621          |
| 115          | 621          | 115       | 621          | 115       | 621          | 115       | 621          | 115       | 621          |
| 120          | 621          | 120       | 621          | 120       | 621          | 120       | 621          | 120       | 621          |
| 125          | 621          | 125       | 621          | 125       | 621          | 125       | 621          | 125       | 621          |
| 130          | 621          | 130       | 621          | 130       | 621          | 130       | 621          | 130       | 621          |
| 135          | 621          | 135       | 621          | 135       | 621          | 135       | 621          | 135       | 621          |
| 140          | 621          | 140       | 621          | 140       | 621          | 140       | 621          | 140       | 621          |
| 145          | 621          | 145       | 621          | 145       | 621          | 145       | 621          | 145       | 621          |
| 150          | 621          | 150       | 621          | 150       | 621          | 150       | 621          | 150       | 621          |
| 155          | 621          | 155       | 621          | 155       | 621          | 155       | 621          | 155       | 621          |
| 160          | 621          | 160       | 621          | 160       | 621          | 160       | 621          | 160       | 621          |
| 165          | 621          | 165       | 621          | 165       | 621          | 165       | 621          | 165       | 621          |
| 170          | 621          | 170       | 621          | 170       | 621          | 170       | 621          | 170       | 621          |
| 175          | 621          | 175       | 621          | 175       | 621          | 175       | 621          | 175       | 621          |
| 180          | 621          | 180       | 621          | 180       | 621          | 180       | 621          | 180       | 621          |
| 180          | 1713         | 180       | 1639         | 180       | 1569         | 180       | 1502         | 180       | 1384         |
| 185          | 1759         | 185       | 1691         | 185       | 1625         | 185       | 1565         | 185       | 1462         |
| 190          | 1811         | 190       | 1747         | 190       | 1688         | 190       | 1635         | 190       | 1549         |
| 195          | 1868         | 195       | 1810         | 195       | 1758         | 195       | 1712         | 195       | 1645         |
| 200          | 1931         | 200       | 1880         | 200       | 1835         | 200       | 1797         | 200       | 1752         |
| 205          | 2000         | 205       | 1957         | 205       | 1920         | 205       | 1892         | 205       | 1871         |
| 210          | 2077         | 210       | 2042         | 210       | 2015         | 210       | 1998         | 210       | 1998         |
| 215          | 2162         | 215       | 2136         | 215       | 2119         | 215       | 2114         | 215       | 2114         |
| 220          | 2256         | 220       | 2240         | 220       | 2235         | 220       | 2235         | 220       | 2235         |
| 225          | 2360         | 225       | 2355         | 225       | 2355         | 225       | 2355         | 225       | 2355         |
| 230          | 2475         | 230       | 2475         | 230       | 2475         | 230       | 2475         | 230       | 2475         |



**TABLE B-4**  
**48 EFPY Cooldown Curve Data Points Using 1998 App. G "Circ-Flaw" Methodology**  
**(w/K<sub>IC</sub>, w/Flange Notch and w/o Uncertainties for Instrumentation Errors)**

| Steady-State |              | -20°F/hr  |              | -40°F/hr  |              | -60°F/hr  |              | -100°F/hr |              |
|--------------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|-----------|--------------|
| Temp (°F)    | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) | Temp (°F) | Press (psig) |
| 60           | 0            | 60        | 0            | 60        | 0            | 60        | 0            | 60        | 0            |
| 60           | 621          | 60        | 621          | 60        | 621          | 60        | 621          | 60        | 621          |
| 65           | 621          | 65        | 621          | 65        | 621          | 65        | 621          | 65        | 621          |
| 70           | 621          | 70        | 621          | 70        | 621          | 70        | 621          | 70        | 621          |
| 75           | 621          | 75        | 621          | 75        | 621          | 75        | 621          | 75        | 621          |
| 80           | 621          | 80        | 621          | 80        | 621          | 80        | 621          | 80        | 621          |
| 85           | 621          | 85        | 621          | 85        | 621          | 85        | 621          | 85        | 621          |
| 90           | 621          | 90        | 621          | 90        | 621          | 90        | 621          | 90        | 621          |
| 95           | 621          | 95        | 621          | 95        | 621          | 95        | 621          | 95        | 621          |
| 100          | 621          | 100       | 621          | 100       | 621          | 100       | 621          | 100       | 621          |
| 105          | 621          | 105       | 621          | 105       | 621          | 105       | 621          | 105       | 621          |
| 110          | 621          | 110       | 621          | 110       | 621          | 110       | 621          | 110       | 621          |
| 115          | 621          | 115       | 621          | 115       | 621          | 115       | 621          | 115       | 621          |
| 120          | 621          | 120       | 621          | 120       | 621          | 120       | 621          | 120       | 621          |
| 125          | 621          | 125       | 621          | 125       | 621          | 125       | 621          | 125       | 621          |
| 130          | 621          | 130       | 621          | 130       | 621          | 130       | 621          | 130       | 621          |
| 135          | 621          | 135       | 621          | 135       | 621          | 135       | 621          | 135       | 621          |
| 140          | 621          | 140       | 621          | 140       | 621          | 140       | 621          | 140       | 621          |
| 145          | 621          | 145       | 621          | 145       | 621          | 145       | 621          | 145       | 621          |
| 150          | 621          | 150       | 621          | 150       | 621          | 150       | 621          | 150       | 621          |
| 155          | 621          | 155       | 621          | 155       | 621          | 155       | 621          | 155       | 621          |
| 160          | 621          | 160       | 621          | 160       | 621          | 160       | 621          | 160       | 621          |
| 165          | 621          | 165       | 621          | 165       | 621          | 165       | 621          | 165       | 621          |
| 170          | 621          | 170       | 621          | 170       | 621          | 170       | 621          | 170       | 621          |
| 175          | 621          | 175       | 621          | 175       | 621          | 175       | 621          | 175       | 621          |
| 180          | 621          | 180       | 621          | 180       | 621          | 180       | 621          | 180       | 621          |
| 180          | 1517         | 180       | 1423         | 180       | 1329         | 180       | 1235         | 180       | 1052         |
| 185          | 1543         | 185       | 1452         | 185       | 1360         | 185       | 1270         | 185       | 1094         |
| 190          | 1572         | 190       | 1483         | 190       | 1395         | 190       | 1308         | 190       | 1142         |
| 195          | 1604         | 195       | 1518         | 195       | 1433         | 195       | 1351         | 195       | 1195         |
| 200          | 1639         | 200       | 1557         | 200       | 1476         | 200       | 1398         | 200       | 1253         |
| 205          | 1678         | 205       | 1600         | 205       | 1523         | 205       | 1450         | 205       | 1319         |
| 210          | 1721         | 210       | 1647         | 210       | 1576         | 210       | 1509         | 210       | 1391         |
| 215          | 1769         | 215       | 1700         | 215       | 1634         | 215       | 1573         | 215       | 1472         |
| 220          | 1822         | 220       | 1758         | 220       | 1698         | 220       | 1645         | 220       | 1561         |
| 225          | 1880         | 225       | 1822         | 225       | 1770         | 225       | 1724         | 225       | 1661         |
| 230          | 1944         | 230       | 1893         | 230       | 1849         | 230       | 1812         | 230       | 1771         |
| 235          | 2015         | 235       | 1972         | 235       | 1936         | 235       | 1910         | 235       | 1894         |
| 240          | 2094         | 240       | 2059         | 240       | 2033         | 240       | 2018         | 240       | 2018         |
| 245          | 2180         | 245       | 2155         | 245       | 2140         | 245       | 2137         | 245       | 2137         |
| 250          | 2276         | 250       | 2262         | 250       | 2259         | 250       | 2259         | 250       | 2259         |
| 255          | 2382         | 255       | 2379         | 255       | 2379         | 255       | 2379         | 255       | 2379         |

**APPENDIX C**  
**Pressurized Thermal Shock (PTS) Results for 48 EFPY**

The 48 EFPY operating term has an associated fluence value for the beltline materials of:

- For the intermediate and lower shell longitudinal welds, Fluence<sub>48 EFPY</sub> =  $1.295 \times 10^{19}$  n/cm<sup>2</sup>
- For the remaining beltline materials, Peak Fluence<sub>48 EFPY</sub> =  $1.906 \times 10^{19}$  n/cm<sup>2</sup>

RT<sub>PTS</sub> predicted values for 48 EFPY are shown in Table C-1.

**Table C-1**  
**RT<sub>PTS</sub> Calculations for Indian Point Unit 2 Beltline Region Materials at 48 EFPY**

| Material  | Fluence<br>(n/cm <sup>2</sup> ,<br>E>1.0 MeV) | FF    | CF<br>(°F) | ΔRT <sub>PTS</sub> <sup>(a)</sup><br>(°F) | Margin<br>(°F) | RT <sub>NDT(U)</sub> <sup>(b)</sup><br>(°F) | RT <sub>PTS</sub> <sup>(c)</sup><br>(°F) |
|---|---|-------|------------|---|----------------|---|--|
| Inter. Shell Plate B-2002-1                               | $1.906 \times 10^{19}$                        | 1.176 | 144        | 169.34                                    | 34             | 34  | 237                                      |
| - Using S/C Data  | $1.906 \times 10^{19}$                        | 1.176 | 114        | 134.06                                    | 17             | 34  | 185                                      |
| Inter. Shell Plate B-2002-2                               | $1.906 \times 10^{19}$                        | 1.176 | 115.1      | 135.36                                    | 34             | 21  | 190                                      |
| - Using S/C Data  | $1.906 \times 10^{19}$                        | 1.176 | 118.2      | 139.00                                    | 34             | 21  | 194                                      |
| Inter. Shell Plate B-2002-3                               | $1.906 \times 10^{19}$                        | 1.176 | 176        | 206.98                                    | 34             | 21  | 262                                      |
| - Using S/C Data  | $1.906 \times 10^{19}$                        | 1.176 | 181.9      | 213.91                                    | 17             | 21  | 252                                      |
| Lower Shell Plate B-2003-1                                | $1.906 \times 10^{19}$                        | 1.176 | 152        | 178.75                                    | 34             | 20  | 233                                      |
| Lower Shell Plate B-2003-2                                | $1.906 \times 10^{19}$                        | 1.176 | 128.8      | 151.53                                    | 34             | -20   | 166                                      |
| Intermediate & Lower Shell<br>Long. Welds (Heat # W5214)  | $1.295 \times 10^{19}$                        | 1.072 | 230.2      | 246.78                                    | 65.5           | -56   | 256                                      |
| - Using S/C Data  | $1.295 \times 10^{19}$                        | 1.072 | 251.8      | 269.9                                     | 44.0           | -56   | 258                                      |
| Intermediate to Lower Shell<br>Girth Weld (Heat # 34B009) | $1.906 \times 10^{19}$                        | 1.176 | 220.9      | 259.78                                    | 65.5           | -56   | 269                                      |

Notes:

(a)  $\Delta RT_{PTS} = CF * FF$

(b) Initial RT<sub>NDT</sub> values are measured values, except for the weld materials.

(c)  $RT_{PTS} = RT_{NDT(U)} + \Delta RT_{PTS} + \text{Margin (°F)}$

### RT<sub>PTS</sub> Assessment Conclusion:

Indian Point Unit 2 vessel materials will remain below the screening criteria for axial welds and plates (270°F) and for circumferential welds (300°F) at 48 EFPY.

**APPENDIX D**  
**Predicted Upper Shelf Energy (USE) Values for 48 EFPY**

Based on the projected fluences detailed in Section 4, the calculated 1/4T values are shown in Table D-1 and the projected USE drops are shown in Table D-2.

**Table D-1**  
**48 EFPY 1/4T Fluence Values for all the Indian Point Unit 2 Beltline Materials**

| Material   | Fluence @ 48 EFPY <sup>(a)</sup> | 1/4T Fluence @ 48 EFPY <sup>(b)</sup> |
|--|----------------------------------|---------------------------------------|
| Intermediate Shell Plate B-2002-1                            | 1.906                            | 1.136                                 |
| Intermediate Shell Plate B-2002-2                            | 1.906                            | 1.136                                 |
| Intermediate Shell Plate B-2002-3                            | 1.906                            | 1.136                                 |
| Lower Shell Plate B-2003-1                                   | 1.906                            | 1.136                                 |
| Lower Shell Plate B-2003-2                                   | 1.906                            | 1.136                                 |
| Intermediate & Lower Shell Longitudinal Welds (Heat # W5214) | 1.295                            | 0.772                                 |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009)       | 1.906                            | 1.136                                 |

Notes:

(a) f @ 48 EFPY is the 48 EFPY fluence at the clad/base metal interface ( $\times 10^{19}$  n/cm<sup>2</sup>, E > 1.0 MeV).

(b) 1/4T f @ 48 EFPY = f @ 48 EFPY \* e<sup>(-0.24\*X)</sup>, where X is the depth into the vessel wall (X = 0.25 \* 8.625 inches = 2.156 inches).

**Table D-2**  
**Predicted 48 EFPY USE Calculations for all the Beltline Region Materials**

| Material   | Weight % of Cu | 1/4T EOL Fluence (10 <sup>19</sup> n/cm <sup>2</sup> ) | Unirradiated USE <sup>(a)</sup> (ft-lb) | Projected USE Decrease (%) | Projected EOL USE (ft-lb) |
|--|----------------|--|---|----------------------------|---------------------------|
| Intermediate Shell Plate B-2002-1                            | 0.19           | 1.136  | 70                                      | 29                         | 49.7                      |
| Intermediate Shell Plate B-2002-2                            | 0.17           | 1.136  | 73                                      | 28                         | 52.6                      |
| Intermediate Shell Plate B-2002-3                            | 0.25           | 1.136  | 74                                      | 35                         | 48.1                      |
| Lower Shell Plate B-2003-1                                   | 0.20           | 1.136  | 71                                      | 30                         | 49.7                      |
| Lower Shell Plate B-2003-2                                   | 0.19           | 1.136  | 88                                      | 29                         | 62.5                      |
| Intermediate & Lower Shell Longitudinal Welds (Heat # W5214) | 0.21           | 0.772  | 121                                     | 33                         | 81.1                      |
| Intermediate to Lower Shell Girth Weld (Heat # 34B009)       | 0.19           | 1.136  | 82                                      | 34                         | 54.1                      |

Notes:

(a) These values were obtained from WCAP-15629, Rev. 1.



**USE Assessment Conclusion:**

Several materials would drop below the 50 ft-lb threshold set by 10 CFR 50 Appendix G. An alternative analysis method would need to be performed to justify continued operation. This was actually done as part of the WOG back in 1993 for all WOG participants. WCAP-13587, Revision 1, "Reactor Vessel Upper Shelf Energy Bounding Evaluation for Westinghouse Pressurized Water Reactors" documents that the minimal acceptable USE value is 43 ft-lbs for 4-loop plants. All of the Indian Point 2 vessel beltline materials exceed this minimum acceptable value. Per NUREG-1511, "Reactor Pressure Vessel Status Report," dated December 1994, licensees wanting to utilize the WCAP-13587 analyses to demonstrate compliance with USE requirements need to make a submittal to the NRC describing applicability of the WCAP to their plant and make a request to the NRC for review and approval.