| Ċ | | RNATIONAL | CALCUL PACKAGE CO | ATION VER SHEET | 71 | culation No: 1 60-2025 [NP] Page 1 | | |
|------------|--|-------------------------------|--|---------------------|----------------------------------|--|--|--|
| | PROJECT N | AME: NEWGEN | | OLIENT. | AC Internatio | onal | | |
| | CALCULATI | ON TITLE: | NONPROPE | RETARY | | ······ | | |
| | Fuel Assembly Structural Evaluation | | | | | | | |
| | PROBLEM STATEMENT OR OBJECTIVE OF THE CALCULATION: | | | | | | | |
| | inventory of | contained in this ca | on for the end drop o alculation. The end di 0 GWD/MTU) of the | rop acceleration is | s limited to 4 | | | |
| | allowing a | A side dro maximum spacing | p evaluation is perfo of 60 inches. | rmed for PWR as | semblies witl | n a missing grid | | |
| | Revision | Affected Pages | Revision Descript | ion of Prep | nd Initials parers & ckers | Functional Manager Approval/Date | | |
| | 0 | 1-15 A1 – A4 B1 – B7 | Initial Issue | Mike | Yaksh | Mike Yaksh 8/05/07 | | |
| | | | | Alar | n Lin | | | |
| | 1 | 1,2,3 B1-B8 C1-C13 | Added side drop evaluation for PW with a missing gr | IR Mikelfab | e Wang | Mike Yaksh Muhu Jakola 8/31/07 | | |
| | | | | | | | | |
| \bigcirc | | | | | | | | |

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INDEPENDENT DESIGN VERIFICATION CHECK SHEET

| Calculation Number: | <u>71160-2025</u> | Revision: | <u>1</u> | |
|--|---------------------|------------------|----------|---|
| Scope of Analysis File: | Fuel assembly struc | tural evaluation | | |
| Review Methodology: Confirm that the Calcul | | es: n): | | |
| 1. Statement of Purpose | | | | / |
| 2 Defined Method of | | | | |

| 2. | | - |
|----|--|-----|
| 3. | Listing of Assumptions | |
| | Detailed Analysis Record | |
| | Statement of Conclusions / Recommendations (if applicable) | |
| | References | r - |

Verification Step Activities Yes N/A Comments No For the scope of the defined analysis: ł A. Are the required data input complete? 1. Material Properties 2. Geometry (Drawing Reference) 3. Loading Source Term If a supporting analysis is required to define the load state, 4. has it been defined? B. Are Boundary conditions acceptable? 2 Is the method of analysis adequate for the defined scope? 3 Is the worst case loading/configuration documented? 4 V Are the acceptance criteria defined and complete? $\overline{\mathsf{v}}$ 5 Has all concurrent loading been considered? 6 Are analyses consistent with previous work for method and approach? 7 V Are the records for input and output complete? 8 Has the computer output been verified? L 9 Is traceability to verified software complete? ν Is the statement of conclusions and recommendations complete and 10 acceptable for the project and objectives of the defined purpose? 11 Are references complete? v 12 Are results reasonable for purpose of calculation? See SP-111 13 Has the cumulative effect of specified dimension tolerances on the fabrication/operation fit-up been addressed?

Christike X War Reviewer (1 8/30/0-7 ne/Signature)

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| 2.0 | PURPOSE | 5 |
| 3.0 | METHOD OF ANALYSIS | 5 |
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APPENDIX AANSYS and LS-DYNA OutputAPPENDIX BANSYS/LS-DYNA Input / Output Files on DiskAPPENDIX CEvaluation of Side Drop for a PWR with a Missing Grid

| | Calc. No71160-2025, Rev. 1 |
|----------------------|----------------------------|
| INTERNATIONAL | Page 3 of 15 |

1.0 SYNOPSIS OF RESULTS

A structural evaluation is performed for the PWR and BWR fuel assembly for end drop conditions. The detailed evaluation is performed for the PWR which is shown to bound the BWR fuel rod.

The PWR and BWR fuel inventory was taken from Ref.1. The table below shows a sample of the fuel assemblies and contains the bounding conditions in terms of the minimum and maximum cross sectional moment of inertia.

| PWR Fuel Assembly | Cladding Diameter (in) | Cladding Thickness (in) | Fuel Rod Pitch (in) | Gap between fuel assembly and fuel tube wall (in) |
|----------------------|------------------------------|-------------------------------|---------------------------|--|
| We 17x17 | 0.360 | 0.021 | 0.496 | 0.564 |
| We 15x15 | 0.417 | 0.024 | 0.563 | 0.561 |
| We 14x14 | 0.400 | 0.022 | 0.556 | 1.232 |
| CE16x16 | 0.382 | 0.025 | 0.506 | 0.888 |
| CE14x14 | 0.440 | 0.026 - 0.031 | 0.580 | 0.880 |
| BW17x17 | 0.377 | 0.022 | 0.502 | 0.451 |
| BW15x15 | 0.414 | 0.022 | 0.568 | 0.494 |

| BWR Fuel Assembly | Cladding Diameter (in) | Cladding Thickness (in) |
|----------------------|------------------------------|-------------------------------|
| GE 7x7 | 0.563 | 0.032 |
| GE 7x7 | 0.563 | 0.032 |
| GE 8x8-2 | 0.483 | 0.032 |
| GE 8x8-2 | 0.483 | 0.032 |
| GE 8x8-4 | 0.484 | 0.032 |
| GE 8x8-4 | 0.484 | 0.032 |
| GE 9x9-2 | 0.441 | 0.028 |
| GE 10x10-2 | 0.378 | 0.024 |

The PWR fuel assembly is considered to have a missing grid at the bounding location near the bottom end of the fuel assembly. The missing grid condition is simulated by considering the distance from the bottom of the fuel rod to the first grid to be 60 inches.

A transient evaluation was performed for the PWR fuel rod for a maximum bowed condition of 1.23 inch. The minimum safety determined for the bounding end drop condition is 2.00.

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2.0 PURPOSE

The purpose of this Calculation is to perform an evaluation of the end drop condition for the fuel assembly.

3.0 METHOD OF ANALYSIS

For the end drop evaluation a transient evaluation is performed using LS-DYNA.

4.0 DESIGN INPUT AND ASSUMPTIONS

4.1 Design Input

The following design input is used for the PWR fuel rod evaluation (Reference 1):

The following design input is used for the BWR fuel rod evaluation (Reference 1):

| Clad and pellet properti | ies: |
|--------------------------|------|
|--------------------------|------|

| | | Modulus of Elasticity | Density (lb/in ³) | |
|----------|-----|-----------------------|-------------------------------|--|
| | | (10 ⁶ psi) | | |
| Rod Cla | ad | 10.47 (Ref. 5) | 0.237 (Ref. 2) | |
| Fuel pel | let | 13 (Ref. 3) | 0.396 (Ref. 3) | |

Dimensions for the PWR basket tube: (maximum slot opening is 8.86 inches) Ref. 4

| N AC | ······································ | Calc. No. 71160-2025, Rev. 0 |
|-------------|--|------------------------------|
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4.2 Assumptions/Design Considerations

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5.0 ANALYSIS DETAILS

Transient analyses using LS-DYNA are performed to evaluate the transient response in the end drop for the PWR fuel rods.

5.1 End Drop Analysis of the PWR fuel rod

In the end drop orientation, the fuel rods are laterally restrained by the grids and come into contact with the fuel assembly base. As opposed to employing a straight fuel assembly in the evaluation with all the grids present, the fuel assembly is considered to have a maximum possible bow with or without a missing grid (the configuration still meets the acceptable configuration for undamaged fuel). The evaluation of the PWR fuel rod is based on the following representative samples of PWR fuel assemblies:

| PWR Fuel Assembly | Cladding Diameter (in) | Cladding Thickness (in) | Fuel Rod Pitch (in) | Gap between fuel assembly and fuel tube wall (in) |
|----------------------|------------------------------|-------------------------------|---------------------------|--|
| We 17x17 | 0.360 | 0.021 | 0.496 | 0.564 |
| We 15x15 | 0.417 | 0.024 | 0.563 | 0.561 |
| We 14x14 | 0.400 | 0.022 | 0.556 | 1.232 |
| CE16x16 | 0.382 | 0.025 | 0.506 | 0.888 |
| CE14x14 | 0.440 | 0.026 - 0.031 | 0.580 | 0.880 |
| BW17x17 | 0.377 | 0.022 | 0.502 | 0.451 |
| BW15x15 | 0.414 | 0.022 | 0.568 | 0.494 |



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Four (4) LS-DYNA models are considered which incorporate the bow of 1.23 inch. These cases envelope the range of the cross sectional moments for the PWR fuel rods and the grid spacing at the bottom of the fuel assembly as summarized in the following Table.

| Case | Lowest grid spacing (inch) | Cross sectional moment of Inertia | Fuel rod OD (inch) | Fuel Clad thickness (w/o oxide effect) (inch) |
|------|-------------------------------|---|-----------------------|--|
| ĺ | 60 | Minimum | 0.36 | 0.021 |
| 2 | 33 | Minimum | 0.36 | 0.021 |
| 3 | 25 | Minimum | 0.36 | 0.021 |
| 4 | 60 | Maximum | 0.44 | 0.031 |

This acceleration

time history is shown below. The velocity associated with this acceleration is 527 in/sec or greater.

| ANAC | | Calc. No. 71160-2025, Rev. 0 |
|------|--|------------------------------|
| | | Page 8 of 15 |

| Time (sec) | Acceleration (in/sec ²) |
|------------|-------------------------------------|
| 0.000 | 0. |
| 0.01 | 10384 |
| 0.02 | 13846 |
| 0.036 | 18461 |
| 0.038 | 15384 |
| 0.045 | 1923 |
| 0.05 | 0 |

The maximum stress intensity at the mid-span of the lowest span of the fuel rod is shown for each case in the table below. The temperature of the fuel at the bottom end of the basket is bounded by $752^{\circ}F$ (400°C) and the static yield strength for irradiated ziracloy at $752^{\circ}F$ is 69.6 ksi (Reference 5).

| Case | Maximum Stress Intensity (ksi) at mid-span of lowest grid spacing | Factor of safety against yield strength |
|------|--|--|
| 1 | 22.8 | 3.05 |
| 2 | 34.8 | 2.00 |
| 3 | 17.0 | 4.09 |
| 4 | 15.2 | 4.58 |

The case using the 33-inch spacing in conjunction with the minimal cross section (Case 2) is identified as the bounding case.



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|------------------------------|--|
| Page 9 of 15 | |

In comparing case 4 to cases

1 and 2, the effect of the maximum cross sectional moment (the ratio of the maximum cross sectional moment to the minimal cross sectional moment is approximately 2.7) indicates that the cross sectional moment has more influence than the grid spacing on the maximum stress.

These results confirm that high burnup PWR fuel with one missing grid will remain undamaged for design basis cask end drop load conditions.

5.2 End Drop Analysis of the BWR fuel rod

The evaluation of the BWR fuel rod is based on the following representative sample of BWR fuel rods from Ref.1:

| Fuel Assembly | Cladding Diameter (in) | Cladding Thickness (in) |
|---------------|------------------------------|-------------------------------|
| GE 7x7 | 0.563 | 0.032 |
| GE 7x7 | 0.563 | 0.032 |
| GE 8x8-2 | 0.483 | 0.032 |
| GE 8x8-2 | 0.483 | 0.032 |
| GE 8x8-4 | 0.484 | 0.032 |
| GE 8x8-4 | 0.484 | 0.032 |
| GE 9x9-2 | 0.441 | 0.028 |
| GE 10x10-2 | 0.378 | 0.024 |

The location of the lateral constraints in the BWR fuel are: 0.00 in, 22.88 in, 43.03 in, 63.18 in, 83.33 in, 103.48 in, 122.3 in, 143.78 in, and 163.42 in.

For the PWR fuel rod (with all grids and with the 120 micron thickness reduction) the largest ratio of unsupported length (L) to radius of gyration of the cladding cross section (r) is

L/r =
$$\frac{33}{0.5 \times \sqrt{((0.360 - 2 \times 0.0047)/2)^2 + (0.318/2)^2}} = 279$$



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The ratio (L/r) for a BWR fuel rod (with the 125 micron thickness reduction for high burnup fuel) is

L/r =
$$\frac{22.88}{0.5 \times \sqrt{((0.378 - 2 \times 0.0049)/2)^2 + (0.330/2)^2}} = 185$$

The analysis presented in Section 5.1 is bounding for both PWR and BWR fuel rods, because the (L/r) for the PWR fuel rod is larger than the (L/r) for the BWR fuel rod. Therefore, no further evaluation of the BWR fuel rod is required.

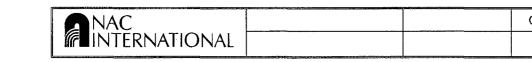
5.3 Input / Output Files

The names of ANSYS macros to generate the models and the LS-DYNA files to perform the LS-DYNA analyses are shown below. The output cover sheets are contained in Appendix A. Listing of the ANSYS macros and LS-DYNA files contained on the compact disk is shown in Appendix B.

| Case ANSYS Macro | | LS-DYAN Input file |
|------------------|-------------------------------|------------------------|
| 1 | posroda.mac | fuelrodbuckling_6a.dyn |
| 2 | posrodb.mac | fuelrodbuckling_6b.dyn |
| 3 | 3 posrodf.mac fuelrodbuckling | |
| 4 | posrodd.mac | fuelrodbuckling_6d.dyn |

6.0 SUMMARY OF RESULTS/CONCLUSIONS

The end drop evaluation of the PWR and BWR fuel assemblies contained in Ref. 1 is performed using bounding parameters. The bounding load condition for the evaluation is the end drop orientation using a maximum acceleration of 48g's. The minimum safety factor for the fuel rod due to bending is 2.0 against the yield strength which confirms that the fuel rod will maintain its integrity and will remain intact during the end drop event.



1.

2.

3.

4.

NAC Drawings:
71160-051 REV. 7, FUEL TUBE ASSEMBLY, TYPE 1, 37 ELEMENT PWR, NEWGEN (as modified by DCR 71160-051-7A)
71160-052 REV. 6, FUEL TUBE ASSEMBLY, TYPE 2, 37 ELEMENT PWR, NEWGEN (as modified by DCR 71160-052-6A)
71160-053 REV. 6, FUEL TUBE ASSEMBLY, TYPE 3, 37 ELEMENT PWR, NEWGEN (as modified by DCR 71160-053-6A)
71160-054 REV. 6, FUEL TUBE ASSEMBLY, TYPE 4, 37 ELEMENT PWR, NEWGEN (as modified by DCR 71160-054-6A)

71160-055 REV. 6, FUEL TUBE ASSEMBLY, TYPE 5, 37 ELEMENT PWR, NEWGEN

5.

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Figures 1, 2 & 3 on pages 13, 14 & 15 are proprietary and, therefore, have been deleted.

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APPENDIX A

ANSYS and LS-DYNA Output

| | | |
|----------------------|------|------------------------------|
| NAC | | Calc. No.: 71160-2025 Rev. 0 |
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COMPUTER OUTPUT COVER SHEET

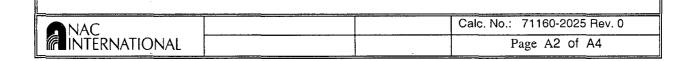
Project Name: MAGNASTOR

Project Number: 71160

File Number (Input File): See Table Below Calculation Number: 71160-2025 Rev. 0 Title of Analysis: ANSYS analyses for the PWR fuel rod – Initial Bow Program: ANSYS Version: 10.0 Originator: ANSYS, Inc. Computer Manufacturer / Operating System: DELL Optiplex GX620/ Windows XP Computer Identification / NAC Number: 00-12-3F-C7-FB-F4 / NAC1550 Hardware Verified: YES X NO (if no, calculation is preliminary) Computer Verification Report Number: EA913-1020-006 Rev. 0 5/3/2006 Output File / Title of Case: See Table Below

Prepared by: MIKE YAKSH Mike Jaka Date: 8/5/07 Checked by: Alan Lin / Remark Date: 8/5/07

| Case | ANSYS Input | ANSYS Output | Date of Run |
|------|-------------|--------------|-------------|
| 1 | posroda.mac | posroda.db | 8/4/07 |
| 2 | posrodb.mac | posrodb.db | 8/4/07 |
| 3 | posrodc.mac | posrodc.db | 8/4/07 |
| 4 | posrodf.mac | posrodf.db | 8/4/07 |



COMPUTER OUTPUT COVER SHEET

| Project Name: MAGNASTOR | |
|--------------------------------|--|
|--------------------------------|--|

Project Number: 71160

File Number (Input File): See Table Below

Calculation Number: 71160-2025 Rev. 0

Title of Analysis: LS-DYNA analyses of the PWR fuel rod enddrop

Version: 970 **Program:** LS-DYNA

Originator: LSTC Inc.

Computer Manufacturer / Operating System: AMD Opteron / Professional X64 Edition Version 2003

Computer Identification / NAC Number: 00-11-25-1E-44-18 / NAC1533

Hardware Verified: YES X NO (if no, calculation is preliminary)

Computer Verification Report Number: EA913-1030-123 Rev. 0

Output File / Title of Case: See Table Below

Date of Run: See Table Below Illike Yaksk, Theke Jebst Prepared by:___ Checked by:

Date:__ Date:

| Case | LS-DYNA MAIN INPUT FILE | D3PLOT file ⁽¹⁾ | Date of D3PLOT file ⁽¹⁾ |
|------|-------------------------|-------------------------------|------------------------------------|
| 2 | FUELRODBUCKLING_6B.DYN | d3plot49 | 8/04/07 |
| 4 | FUELRODBUCKLING_6C.DYN | d3plot48 | 8/05/07 |

(1) This is the last D3PLOT file generated



| ٨C | | Calc. No.: 71160-2025 Rev. 0 |
|--------------------|--|------------------------------|
| TERNATIONAL | | Page A3 of A4 |

COMPUTER OUTPUT COVER SHEET

| Project Name: | MAGNASTOR |
|---------------|-----------|
|---------------|-----------|

Project Number: 71160

File Number (Input File): See Table Below

Calculation Number: 71160-2025 Rev. 0

Title of Analysis: LS-DYNA analyses of the PWR fuel rod enddrop

Program: LS-DYNA Version: 970 Originator: LSTC Inc.

Computer Manufacturer / Operating System: AMD Opteron / Professional X64 Edition Version 2003

Computer Identification / NAC Number: 00-11-25-1E-44-08 / NAC1534

Hardware Verified: YES X NO (if no, calculation is preliminary)

Computer Verification Report Number: EA913-1030-124 Rev. 0

Output File / Title of Case: See Table Below

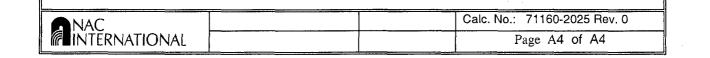
Date of Run: See Table Below MAKE YAKSH Prepared by: Adal Checked by: Alan Lin

Date:_ 8/5/07

Date: 8/5/07

| Case | LS-DYNA MAIN INPUT FILE | D3PLOT file ⁽¹⁾ | Date of D3PLOT file ⁽¹⁾ |
|------|-------------------------|-------------------------------|------------------------------------|
| 1 | FUELRODBUCKLING_6A.DYN | d3plot49 | 8/04/07 |
| 3 | FUELRODBUCKLING_6F.DYN | d3plot49 | 8/05/07 |

(1) This is the last D3PLOT file generated



Appendix B

ANSYS/LS-DYNA Input / Output Files on Disk

Pages B2 through B8 are proprietary and, therefore, have been deleted.

| NAC | Calc. No. 71160-2025 Rev. 1 |
|---------------|-----------------------------|
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APPENDIX C

Evaluation of Side Drop for a PWR with a Missing Grid

| | | |
|------------------------|------|------------------------------|
| ANAC . | | Calc. No.: 71160-2025 Rev. 1 |
| A NAC INTERNATIONAL | | Page C1 of C13 |

C.1 SYNOPSIS OF RESULTS

A structural evaluation is performed for the PWR fuel assembly with a missing grid of up to 60 inches for a 60g side drop condition. The yield strength is the stress allowable. The minimum margin of safety (M.S.) is computed to be +0.45.

| Case | Maximum Stress (ksi) | M.S Against Yield Strength |
|---------|----------------------|----------------------------------|
| CE14×14 | 37.1 | +0.88 |
| WE15×15 | 48.1 | +0.45 |
| WE17×17 | 46.3 | +0.50 |

This confirms that the PWR fuel rod with a missing grid subject to high burn up conditions will remain intact for a side drop condition, which bounds the tip-over accident condition.

C.2 PURPOSE

The purpose of this Appendix is to perform an evaluation of the side drop condition for the PWR fuel assembly with a missing grid.

C.3 METHOD OF ANALYSIS

For the side drop evaluation a static evaluation is performed using ANSYS with a lateral load of 60g's.

C.4 DESIGN INPUT AND ASSUMPTIONS

C.4.1 Design input

The fuel rod dimension and material properties used in this Appendix are taken from Section 4 of the main body of this calculation.

C.4.2 Assumptions/Design Considerations



| Calc. No.: 71160-2025 Rev. 1 |
|------------------------------|
| Page C2 of C13 |

C.5 ANALYSIS DETAILS

Static analyses using ANSYS are performed to evaluate the response in the side drop for the PWR fuel rods with a missing grid, of which the maximum spacing is 60 inches. The analyzed bounding fuel rod length of 60.0 inches envelopes all fuel types and includes the condition with a missing support grid in the fuel assembly. This configuration is evaluated for a 60g side drop. During a side drop, the maximum deflection of a fuel rod is based on the fuel rod spacing of the fuel assembly. Assuming a 17×17 array

, the maximum fuel rod deflection, including the 120 micron oxide layer, is:

| Case | Rod diameter (inches) | Clad thickness (inches) | Cross sectional Area (A) (in ²) | Cross sectional moment (I) (in ⁴) 10 ⁻⁴ | Z (in ³) 10 ⁻³ | Span (inches) Ref. 2 [page] |
|---------|-----------------------------|-------------------------------|--|--|--|--------------------------------|
| CE14×14 | 0.440 | 0.031 | 0.0334 | 6.854 | 3.18 | 16.8[2A-64] |
| WE15×15 | 0.417 | 0.024 | 0.0197 | 2.723 | 2.20 | 26.2 [2A-322] |
| WE17×17 | 0.360 | 0.0205 | 0.0163 | 2.289 | 1.33 | 20.6[2A-346] |

ANSYS is used to perform a static analysis with a lateral loading of 60g.

| L | |
|----------------------|------------------------------|
| | Calc. No.: 71160-2025 Rev. 1 |
| INTERNATIONAL | Page C3 of C13 |

C.5.2 Input / Output Files

The names of ANSYS macros to generate the models and to perform the static analyses are shown below. The table also contains the summary file showing the maximum stress in the beam elements (parameter name STR_MX). The output cover sheets are contained in Section C.6. Listing of the ANSYS macros contained on the compact disk is shown in C.7

| Case | Fuel | ANSYS Macro (*.mac) | ANSYS Summary file (*.sum) |
|------|---------|---------------------|----------------------------|
| 1 | CE14×14 | posrCE_E,,60* | posrCE_E |
| 2 | WE15×15 | posrWE_Y,,60* | posrWE_Y |
| 3 | WE17×17 | posrWE_z,,60* | posrWE_z |

* The second argument is the acceleration (g)

C.6 SUMMARY OF RESULTS/CONCLUSIONS

The side drop evaluation of the PWR fuel assemblies contained in Ref. 1 is performed using a bounding 60 inch spacing for the missing grid. The bounding load condition for the evaluation is the side drop orientation using a maximum acceleration of 60g's. The safety factors for the fuel rod due to the lateral loading is shown in the Table below.

| Case | Maximum Stress (ksi) | M.S Against Yield Strength |
|---------|----------------------|----------------------------------|
| CE14×14 | 37.1 | +0.88 |
| WE15×15 | 48.1 | +0.45 |
| WE17×17 | 46.3 | +0.50 |

The yield strength is the stress allowable. The minimum margin of safety (M.S.) is computed to be +0.45 which confirms that the fuel rod will maintain its integrity and will remain intact during the end drop event.



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Figure C-1.

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C.7 COMPUTER OUTPUT COVER SHEET

Project Name: Magnastor

Project Number: 71160

File Number (Input File): See Table Below

Calculation Number: 71160-2025 Rev. 1

Title of Analysis: ANSYS analyses for the PWR fuel rod - Side drop with a missing grid

Program: ANSYS Version: 10.0 Originator: ANSYS, Inc.

Computer Manufacturer / Operating System: DELL Optiplex GX620/ Windows XP

Computer Identification / NAC Number: 00-12-3F-C7-FB-F4 / NAC1550

Hardware Verified: YES X NO (if no, calculation is preliminary)

Computer Verification Report Number: EA913-1020-006 Rev. 0 5/3/2006

Output File / Title of Case: See Table Below

Date of Run: See Table Below Prepared by: Mike Jack Mike Jack Christine X. Wang Checked by:

Date: 5730/0-Date:_8/30

| Case | Fuel | ANSYS Macro (*.mac) | ANSYS Summary file (*.sum) | Date of Run |
|------|---------|------------------------|-------------------------------|-------------|
| 1 | CE14×14 | posrCE_E | posrCE_E | 8/28/07 |
| 2 | WE15×15 | posrWE_Y | posrWE_Y | 8/29/07 |
| 3 | WE17×17 | posrWE_z | posrWE_z | 8/28/07 |

| NAC INTERNATIONAL | Calc. N | o.: 71160-2025 Rev. 1 |
|-----------------------------|---------|-----------------------|
| | | Page C6 of C13 |

C.8, Listing of ANSYS Macros, on pages C7 through C13 is proprietary and, therefore, has been deleted

C.8 Listing of ANSYS Macros

| | | |
|-----------------------|------|------------------------------|
| ANAC INTERNATIONAL | | Calc. No.: 71160-2025 Rev. 1 |
| | | Page C7 of C13 |