## **ENCLOSURE 3**

MPR Associates, Inc. Technical Report No. MPR-2705, Revision 1, Dated September 2004

"Emergency Diesel Generator Fan & Radiator Performance Evaluation"



MPR-2705 Revision 1 September 2004

# *Emergency Diesel Generator Fan & Radiator Performance Evaluation*

Prepared for

Oyster Creek Nuclear Station Amergen Energy Co., LLC Forked River, NJ 08731



# **Emergency Diesel Generator Fan & Radiator Performance Evaluation**

MPR-2705 Revision 1

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### **Executive Summary**

To assist Amergen Energy evaluate the EDG #1 loose fan drive shaft bearing pillow block event on May 17, 2004 at Oyster Creek Nuclear Station, MPR Associates completed a review of event reports and the Joliet, Illinois MP36 diesel generator test plan and test results. MPR also prepared fan belt slippage calculations and diesel engine jacket water and lube oil temperature rise as a result of reduced air flow through the radiator.

Based on:

- the reported good condition of the fan belt and its subsequent reinstallation on EDG #1;
- the lack of extensive belt squealing noise as contrasted to the banging noise from the pillow block hitting the support pedestal during the event, and;
- most importantly, our analysis of the effects of drive shaft movement during the event;

we conclude that during the time the pillow block was loose and fan belt tension was reduced the EDG #1 fan speed was approximately 573 rpm as compared to the design rated speed of 637 rpm. The lower fan speed (90% of design fan speed) results in a cooling air flow rate through the engine's radiator of 112,400 SCFM as compared to a design air flow rate of 125,000 SCFM.

Further, our analysis concludes the fan belt would have had a belt life of nearly one month if it had been required to continue to operate with slippage at 573 rpm fan speed. Note that as the belt approached its end of life, fan speed would likely degrade further.

The resulting air flow of 112,400 SCFM (90% of design air flow) is sufficient to ensure the EDG #1 EMD Model 645 20-cylinder diesel engine will not overheat when operating at a steady-state load of 2600 kWe. With the ambient air temperature of 70°F, the resulting steady-state engine inlet lube oil temperature is approximately 206°F and the inlet jacket water temperature is approximately 187°F. The plots of engine heat up rate for fan speeds ranging from 100% to 55% of design rated speed are provided in Appendix B.

MPR Associates conclude that the EDG #1 operation would not have been limited by increasing engine jacket water and lube oil temperatures. EDG #1 would not have prematurely been shut down by a high jacket water temperature trip.

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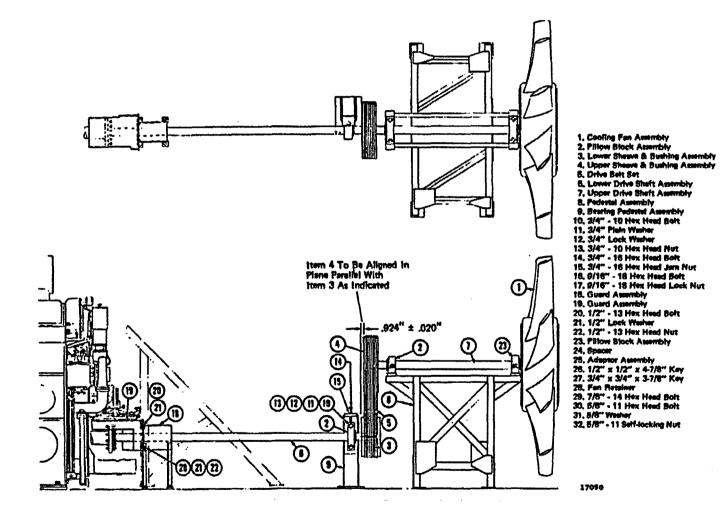
Figure 1-1 MP45A Cooling Fan and Drive Assembly
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#### 1.1 EDG #1 COOLING FAN EVENT BACKGROUND

During a normally scheduled monthly surveillance run of EDG #1 at Oyster Creek Nuclear Station early on May 17, 2004, the plant operating personnel reported hearing an unusual noise coming from the EDG room. The noise was not noticed when the surveillance test began more than an hour earlier. The operator reported a banging noise was coming from the radiator fan drive shaft and belt area. After operating a total of 1 hour and 23 minutes on May 17<sup>th</sup>, EDG #1 was shutdown (Ref. 1). Inspection of the fan drive assembly determined the pillow block assembly (Item 2 on Figure 1-1) that supports the lower drive shaft from the front of the diesel engine was loose.

EDG #1 at Oyster Creek Nuclear Station is known as an MP45A diesel generator set and consists of a 20-cylinder EMD Model 645 diesel engine driving a generator with a design rating of 2,600 kWe. As shown in Figure 1-1, the MP45A is cooled by a radiator with a belt driven fan to exhaust heat from the engine's jacket water system to the atmosphere. The pillow block is mounted to its support pedestal by two <sup>3</sup>/<sub>4</sub>"-10 hex head bolts, nuts and washers (Ref. 2). The lower bolt, washer and nut were found on the floor and the upper bolt was loose. The nut had backed off about 5 full turns or a distance of <sup>1</sup>/<sub>4</sub>". As noted in Reference 1, the fan belt had been replaced during a scheduled maintenance outage and EDG #1 had been returned to service on April 30, 2004. From the time the fan belt was replaced to the start of the surveillance run on May 17, 2004, EDG #1 had operated at idle speed or rated speed for approximately 7 <sup>1</sup>/<sub>4</sub> hours (Ref. 1). The USNRC inspection report (Ref. 3) stated that the event was the result of failure to implement appropriate procedural requirements for maintenance in that "Technicians failed to follow written procedures to torque the cooling fan shaft bearing bolts following fan belt replacement as prescribed by Technical Specification 6.8.1." (From Reference 2)

# Figure 1-1 MP45A Cooling Fan and Drive Assembly



1-2

#### 1.2 OFF-SITE DIESEL ENGINE TESTING TO DEMONSTRATE EDG OPERABILITY

In an attempt to determine the effect of a loosened drive shaft support bearing pillow block, a test was conducted using a similar diesel generator set. A test plan (Ref. 4) to simulate the degraded shaft support condition of EDG #1 was developed by Engine Systems, Inc. (ESI) the nuclear power industry distributor for EMD diesel generators. The test was performed on a similar EMD diesel generator at a power plant in Joliet, Illinois. The diesel generator at Joliet is an MP36 unit which has a 16-cylinder EMD 645 diesel engine as the prime mover. It is radiator cooled with most of the fan shaft support components being similar, if not identical, to the components in the MP45A at Oyster Creek. The differences in fan components between the two units are summarized in Reference 5. The most significant differences are the fan belt sheave diameters, both upper and lower, in the MP36 unit are smaller than used in the MP45A. Thus while both engines having an operating speed of 900 rpm, the resulting fan speed (571 rpm) of the MP36 is slower than the MP45A (637 rpm). The fan belts are the same Goodyear part.

As described in Reference 6, the loose pillow block test completed on the MP36 diesel engine was partially successful. The test of the MP36 diesel generator was also recorded on video tape (Ref.7). A number of interesting observations can be developed from the report and the video tape. Specifically, even with degraded support condition for the lower fan shaft bearing pillow block, the following conditions were demonstrated by the test:

- More than six hours of operation were achieved before the engine shut down due to high jacket water inlet temperature (> 195-200°F).
- The ¾" upper bolt and nut were observed to be erratically oscillating during the test, but they never backed completely out of the pillow block. The pillow block continuously pivoted on the end of the fan belt tension adjusting bolt (Item 14 on Figure 1-1) during the test.
- The fan belt lost a lot of tension during the pillow block movement and began slipping and skipping such that the fan speed degraded to about 56% of design speed (Ref. 6).
- The fan speed became very erratic in the later stages of the test, but the fan never stopped.

#### **1.3 ANALYSES OF DIESEL ENGINE OPERATION AT REDUCED FAN SPEED**

As another method of considering the ability of the EDG #1 to operate for an extended period of time with a loosened drive shaft bearing pillow block, Amergen Energy tasked MPR Associates to calculate the extent of fan belt slippage and the resulting change in fan speed of the EDG. From the reduced airflow that results from reduced fan speed, MPR also calculated the rate at which the EMD 645 20-cylinder diesel engine jacket water and lube oil temperatures would increase. As discussed in Section 2.2, the fan belt was skipping and jumping so much on the MP36 at Joliet as to not be representative of what was likely the case at Oyster Creek. Thus, MPR calculated belt slippage, fan speed and engine heat up rate for EDG #1.

#### 2.1 INDEPENDENT TECHNICAL EVALUATION

As tasked, MPR Associates performed an independent evaluation of the reports related to and the information available from the EDG #1 loose fan drive shaft bearing pillow block event on May 17, 2004 at Oyster Creek Nuclear Station. Our review of the technical information was necessary to obtain much of the design information and operating data associated with the MP45A diesel generator as compared to the MP36 diesel generator used in the test at Joliet, Illinois.

Once we had a good understanding of the facts from those sources, MPR prepared two calculations:

- to determine an estimate of the fan speed resulting from the loose bearing pillow block, and
- to determine the steady-state jacket water and lube oil temperatures within the 20-cylinder EMD Model 645 diesel engine operating at rated load of 2,600 kWe with an ambient air temperature of 70°F.

Section 2.2 provides a summary of the major factors affecting the MP45A fan performance based on our review of the available information. Section 2.3 summarizes assumptions made in the fan speed calculation. Section 2.4 summarizes the results of MPR's calculations of fan speed resulting from the loose pillow block and the resulting EMD 645 diesel engine jacket water and lube oil temperatures when operating at a reduced fan speed. Finally, Appendix A contains the completed EDG #1 fan speed calculations and Appendix B contains the EDG #1 heat up rate calculations.

#### 2.2 FACTORS RELEVANT TO EDG #1 OPERATION

From the technical reports and data associated with the loose pillow block event, there are a number of significant factors or conditions related to the operation of EDG #1 that suggest it would have been capable of operating for an extended period of time without suffering any damage or unusual wear. The following is a brief summary of those factors:

- 1. Even after the Oyster Creek EDG operators reported unusual noises during the May 17<sup>th</sup> surveillance run, EDG #1 completed more than one hour of operation. The operators were able to collect standard log sheet data before shutting down the EDG. All recorded diesel engine operating parameters were "normal" at the time of shut down (Ref. 5).
- 2. After EDG #1 stopped, the reported belt deflection was about 3/8 inch when pushed by a person's fingers (Ref. 5) versus the design value of 7/16 inch when a force of 80 to 104 lbs is applied. Assuming a person can push on the belt with a 45 lb force, that 3/8 inch

deflection suggests the belt tension was approximately 800 lbs. This is slightly less than the calculated belt tension required to drive the fan at its rated speed of 637 rpm. This suggests the EDG #1 fan belt was not slipping much during the time the upper pillow block bolt was loose and the lower bolt was missing.

- 3. It was also reported that the lower drive shaft could not be moved laterally back against the support pedestal without extreme force (Ref 5). This also suggests the fan belt was still under significant tension.
- 4. When EDG #1 was being restored to service on May 17th, mechanics re-tensioned the fan belt, torqued all the pillow block mounting bolts, and took vibrations readings while the engine was operating. No abnormal vibration readings were recorded and EDG #1 was restored to service (Ref. 1).
- 5. The Goodyear fan belt was newly installed on EDG #1 during the planned maintenance outage in April 2004. According to Reference 1, the fan belt had supported EDG#1 for about 10 ½ hours of operation through May 17<sup>th</sup>. The belt was inspected and found to be in good condition (no indication of stretch or burning or overheating). The fan belt was reinstalled on EDG #1 (Ref. 3). This is in contrast to the fan belt from the Joliet engine test which showed signs of glazing on the running surfaces (an indication of excessive slipping) and one edge showed signs of distress in some fibers. The belt actually had fewer hours of operating time during the tests than the belt from EDG #1.
- 6. There was a measured difference in the centerline alignment of the upper and lower drive sheaves on the MP36 test engine in Joliet, Illinois as compared to EDG #1 at Oyster Creek (Ref.5). These minor differences made during generator set field fabrication may explain some of the greater movement of the fan belt and lower drive sheave during the MP36 test.
- 7. On EDG #1 the radiator exhaust louvers regularly cycle from closed to open during normal operation to hold the jacket water coolant nominal temperature at 175°F. The changing louver position results in a cyclic torque load on the drive shaft and fan belt since the fan torque reduces to a very small value when the louvers are closed. However, when the engine jacket water temperature exceeds 175°F, in our calculations (Appendix A), we assume the louvers remain fully open and the fan shaft torque remains stable.
- 8. The lower drive sheave on the MP45A EDG is larger in diameter (15 inch versus 12.5 inch) and therefore heavier than the drive sheave on the MP36 diesel generator set in Joliet, Illinois. This greater weight would tend to increase the tension on the fan belt on EDG #1 and likely helped to further stabilize and maintain the belt tension when the pillow block was loose.
- 9. When the 16-cylinder engine at Joliet was starting to overheat toward the end of the 6-hour run/test due to the extensive fan belt slipping/skipping, the jacket water temperature was observed to increase at the rate of about 1°F per minute (Ref. 6). This is a very rapid rate of temperature rise after once achieving steady-state conditions that was never observed at Oyster Creek.

- 10. According to Reference 6, during initial test setup, the pillow block on the MP36 in Joliet, Illinois lost contact with the belt tensioning bolt (Item 14 on Figure 1-1) when the engine was shut down. The fan belt tensioning was again adjusted and during the formal test the pillow block was always bearing on the belt tensioning bolt. This appears to be the case with EDG #1, namely that the tensioning bolt always remained in contact with the top of the pillow block.
- 11. When the pillow block mounting bolts were both loose, but the lower one still providing some restraining force, there was no reduction of fan speed during the MP36 test at Joliet, Illinois (Ref. 6). It is reasonable to expect that the same condition existed in EDG #1 at Oyster Creek before the lower bolt dropped out on May 17<sup>th</sup>.

Based on the above factors, MPR concludes the drive shaft bearing pillow block movement on EDG #1 was less than occurred during the test of the MP36 diesel generator set.

#### 2.3 CALCULATION ASSUMPTIONS

MPR had to make a number of assumptions in the calculations we prepared to determine fan speed, air flow rate and engine heat up rate and final operating temperatures. Most of these assumptions had to be made because much of the required design data was not available from the following sources:

- General Motors Model 645 diesel engine technical manual for MP45A diesel generator sets
- Engine Systems Incorporated, the nuclear power industry equipment distributor
- The Young Radiator Company, the lube oil cooler and radiator manufacturer
- Koppers, presumed to be the fan manufacturer

Much of the design data needed to very accurately calculate fan and engine operating conditions was either unavailable or considered to be proprietary. Note that MPR did obtain useful technical information from the Goodyear Company, the fan belt manufacturer.

Some specific assumptions made in the fan speed calculations of conditions for EDG #1 include:

- At the fan's rotating speed of 637 rpm, the assumed air flow rate is 125,000 SCFM through the radiator (from Ref. 5).
- As noted above, as the engine heats up above its normal operating temperature, we assumed the louvers on the radiator air flow discharge remained fully open.
- To consider likely ambient conditions at Oyster Creek Nuclear Station during April and May, we assumed the ambient air temperature was 70°F.
- Based on the inspection of the EDG #1 fan belt after the event (Ref. 6), we assumed there was no measurable residual belt stretch.
- Belt tension loss due to centrifugal force as it passes the sheaves is negligible. (This assumption was subsequently confirmed by another calculation.)
- The coefficient of friction between the belt and the sheaves is constant.

- The impact of alignment differences of the upper and lower sheaves on fan belt tension is negligible.
- The fan belt was installed and tensioned in accordance with the guidance in Reference 2 except that the pillow block mounting bolts were not properly torqued. The deflection force assumed on the belt was in the middle of the specified/acceptable range.

The detailed list of assumptions made regarding the EMD Model 645 diesel engine heat up rate are listed on Pages 4 and 5 of the calculation, Appendix B.

#### 2.4 SUMMARY OF CALCULATION RESULTS

#### 2.4.1 Calculated Fan Speed

Using the assumptions noted in Section 2.3, fan belt and mechanical design handbooks, and technical information obtained from Goodyear, MPR calculated the fan speed for EDG #1 as a result of the loss of belt tension due to the loose pillow block. As shown in Appendix A, the calculated fan speed during the EDG #1 surveillance run on May 17, 2004 was 573 rpm. The design fan speed is 637 rpm, thus the calculated 573 rpm represents a 10% reduction. The axial fan performance at 90% of design rated speed is determined to be about 90% of its design value or approximately 112,400 SCFM. Finally, we calculated the fan belt operating under these conditions would have an estimated service life of nearly one month of continuous operation.

#### 2.4.2 Calculated EMD Diesel Engine Heat Up Rate

As demonstrated during the Joliet test, a fan speed of 307 to 320 rpm is unacceptable in that the 16-cylinder EMD 645 diesel engine overheated after about six hours. However, at a calculated fan speed of 573 rpm versus the design speed of 637 rpm, the EDG #1 fan is still providing sufficient air flow (approximately 90% of design air flow) through the radiator that the 20-cylinder EMD 645 diesel engine operating at rated load <u>will not</u> overheat. Based on our calculations in Appendix B, assuming an ambient temperature of 70°F, the engine jacket water inlet temperature reaches a steady-state value of approximately 187°F and the engine lube oil inlet temperature reaches a steady-state value of approximately 206°F. Both of these temperatures are higher than measured for EDG #1 (Ref. 8), however they are lower than the EMD Model 645 diesel engine shutdown temperature limits (Ref. 9).

The maximum steady-state engine temperature conditions were also calculated at an ambient air temperature of 70°F to ensure the calculations in Appendices A and B accurately represented the actual conditions before the fan belt loss-of-tension event. With the fan running at the design rated speed of 637 rpm, the calculated maximum engine jacket water inlet temperature is 181°F and the maximum engine lube oil temperature is 196°F. On April 30, 2004, the recorded ambient temperature was 70°F, thus the recorded data from that date shows very good agreement, within 3°F, with the calculated results in Table 2-1 below. As shown in Table 2-1, these temperatures in Appendix B. There is only one area where we have made an adjustment in the results and that is regarding lube oil temperature rise within the engine. Based on

information provided in Reference 10, we were provided an estimated ratio between the engine heat discharged through the radiator and the heat discharged through the lube oil cooler. This guideline information is for sizing radiators and coolers for a number of engines and is an estimate. As noted in Appendix B, our calculated engine lube oil temperature rise is 8.6°F more than the maximum actually measured during six different EDG #1 surveillance runs recorded in Reference 8. These recorded data cover EDG #1 surveillance runs with ambient air temperatures ranging from 32°F to 70°F. In all six cases with the engine operating at rated load, the recorded lube oil temperature rise within the engine is between 11°F and 15°F. Thus, we have included that lube oil temperature rise of 15°F in the results summarized below. This adjustment does not have a significant effect on the calculated engine heat up rate and the inlet jacket water and lube oil temperatures calculated in Appendix B.

Engine Operating Conditions (Rated Load, 2,600 kWe)	Air Flow Conditions Across Radiator	Engine Jacket Water Inlet Temp, °F	Engine Jacket Water Outlet Temp, °F	Engine Lube Oil Inlet Temp, °F	Engine Lube Oil Outlet Temp, °F
Ambient air temperature, 70°F (on April 30, 2004; from Ref. 8)	Actual flow during the EDG #1 surveillance run <b>before</b> bolts became loose.	178	 Not recorded	194	208
Ambient air temperature, 70°F	Calculated at 100% design flow rate; 637 rpm (Appendix B)	181	190.5	196	211
Ambient air temperature, 70°	Calculated at 90% design flow rate; 573 rpm (Appendix B)	187	196	204	219
Ambient air temperature, 70°F	Calculated at 55% design flow rate; 350 rpm (Appendix B)	223	232.5	239	254

#### Table 2-1 Summary of Actual and Calculated Diesel Engine Operating Temperatures

# **3** Conclusions

The Joliet MP36 degraded fan drive shaft bearing support test served the purpose of bounding the event and suggesting that even in the degraded condition the engine was able to operate for more than six hours without over heating.

Based on our review of event reports, the MP36 test plan and test results, and our independent fan belt slippage calculations, we conclude that the fan speed would have been reduced during the EDG #1 surveillance test on May 17, 2004. Based on:

- the reported good condition of the fan belt and its subsequent reinstallation on EDG #1;
- the lack of extensive belt squealing noise as contrasted to the banging noise from the pillow block hitting the support pedestal during the event; and;
- most importantly, our analysis of the effects of drive shaft movement during the event;

we conclude that during the time the pillow block was loose and fan belt tension was reduced the EDG #1 fan speed was approximately 573 rpm as compared to the design rated speed of 637 rpm. The lower fan speed (90% of design fan speed) results in a cooling air flow rate through the engine's radiator of 112,400 SCFM as compared to a design air flow rate of 125,000 SCFM.

Further, our analysis concludes the fan belt would have had a belt life of nearly one month if it had been required to continue to operate with slippage at 573 rpm fan speed. Note that as the belt approached its end of life, fan speed would likely degrade further.

The resulting air flow of 112,400 SCFM (90% of design air flow) is sufficient to ensure the EDG #1 EMD Model 645 20-cylinder diesel engine will not overheat when operating at steady-state load of 2600 kWe. With the ambient air temperature of 70°F, the resulting steady-state lube oil inlet temperature is approximately 206°F and the jacket water inlet temperature is approximately 187°F. The plots of EMD Model 645 diesel engine heat up rate and steady-state temperatures for fan speeds ranging from 100% to 55% of design rated speed are provided in Appendix B.

MPR Associates conclude that the EDG #1 operation would not have been limited by increasing engine jacket water and lube oil temperatures. EDG #1 would not have prematurely been shut down by a high jacket water temperature trip.



- 1. Prompt Investigation Report, Oyster Creek, subject: "Failure of The #1 EDG Cooling Fan," date of event May 17, 2004.
- 2. General Motors Electro-Motive Division Maintenance Instruction 1200, Rev A, dated February 1979, subject: "MP45 Cooling Fan and Related Drive Train Assembly."
- USNRC Report EA-04-142 dated August 12, 2004, subject: "Oyster Creek Generating Station – NRC Inspection Report 050000219/2004003; Preliminary Greater Than Green Finding."
- 4. ESI Document Number 6012458-TP-1 Revision 0 July 6, 2004, Subject: "Test Plan of an EMD MP Radiator Fan Drive with Degraded Lower Pillow Block Bearing Mounting Bolts."
- 5. "Oyster Creek EDG Cooling Fan Drive Test, Technical Background and Basis," forwarded by e-mail dated August 18, 2004.
- 6. ESI Document Number 6012458-TR-1 Revision 1 August 25, 2004, Subject: "Test Report of an EMD MP Radiator Fan Drive with Degraded Lower Pillow Block Bearing Mounting Bolts."
- 7. Unified Engineering, Inc. Video Tape dated August 2, 2004, subject: "Condensed Video of all Bracket Testing at Joliet Station on 7/28/04 and 7/29/04."
- 8. E-mail from Amergen Energy (Dave Jones) to MPR (Art Killinger) on August 27, 2004, subject: Log Sheet Data of EDG #1 Surveillance Parameters.
- 9. General Motors Electro-Motive Division Application and Installation Data, GM Series 645 Diesel Power Units dated December 1983 (data for turbocharged 16- and 20-cylinder engines).
- 10. Memorandum of Telecon between MPR and Exelon, Subject: EMD Model 645 Diesel Engine Performance Details/Guidelines, dated August 25, 2004.



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Title: Engine Driven Radiator	Fan Speed Calculation			alculation No. 083-0314-CZ
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#### 1.0 PURPOSE

The purpose of this calculation is to determine the rotational speed of the engine driven radiator fan on EDG No. 1 at Oyster Creek Nuclear Generating Station. Following maintenance work, the radiator fan drive belt tension degraded because the mounting bolts on the fan drive shaft pillow block assembly were not properly installed. The lower bolt fell out and the upper bolt loosened during a monthly surveillance test. The calculated fan speed will be used to determine the radiator fan air flow rate which will be used in an evaluation of the EDG system temperatures. The service life of the belt in the degraded tension condition is also estimated.

#### 2.0 SUMMARY OF RESULTS

The calculated fan speed is 573 rpm. The air flow rate provided by the fan is 112,400 cfm. The estimated belt life in the degraded tension condition is 618 hours.

#### 3.0 DISCUSSION

#### 3.1 Assumptions

- 1. Belt does not stretch during the event. The heating and stretch of the Goodyear Flexten belt is considered minimal during the period of time that the engine needs to run (maximum stretch of the belt is 1.5% of its original length over the 99,000-hour design life of the belt, References 7 and 1).
- 2. Tension due to centrifugal force is negligible.
- 3. Coefficient of friction between belt and sheaves is constant.
- 4. The belt was installed and tensioned in accordance with the guidance in Reference 5 except that the pillow block bolts were not tightened. The deflection force applied was in the middle of the specified range.
- 5. The impact of misalignment of the upper and lower sheaves on belt pretension is negligible. The effect of vertical alignment of the sheaves and the lateral alignment of the lower fan shaft on the belt slippage is included in the center distance calculation.

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#### 3.2 Parameters of Radiator Fan Belt Drive for Oyster Creek EDG No. 1

The parameters of the belt drive for Oyster Creek EDG No. 1 are summarized in Table 1.

Param	References	
Lower Sheave Effective Diameter	d = 15.0"	Reference 8
Upper Sheave Effective Diameter	D = 21.2"	Reference 8
Fan Belt Assembly	Torque Team Plus 5VF1120	Reference 9
Effective Belt Length	L = 112"	Reference 1, 8
Driving Shaft Rated Speed	N <sub>1</sub> = 900 rpm	Reference 8
Fan Rated Speed	N <sub>2</sub> = 637 rpm	Reference 8
Number of Ribs	N = 8	Reference 1
Maximum Deflection Force by GM	$F_L = 13 \times 8 = 104 \text{ lbs}$	Reference 5
Belt Modulus Factor	K = 4.88	Reference 2
Belt Weight/belt or rib	W = 0.16 lb/ft	Reference 2

Table 1. Belt Drive Parameter for MP45A

#### **3.3** Torque and Tension Equation Derivation

The torque transmitted by the belt is:

 $\tau = D/2 (T_1 - T_2) = D/2 \cdot T_e$ (Reference 3)

where:

D = driven sheave effective diameter, in = 21.2 in  $T_e$  = effective belt tension, lbs =  $(T_1 - T_2)$  $T_1$  = tight side belt tension, lbs  $T_2$ = slack side belt tension, lbs

For the same belt drive in two different tension conditions, the torque transmitted by the drive is directly proportional to the effective tension. Therefore,

$$\frac{\tau_1}{\tau_2} = \frac{T_{e1}}{T_{e2}}$$
(Eq. 1)

Installation tension:  $T_i = (T_1 + T_2)/2$ (Reference 3) (Eq. 2)

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N = number of belts	and tension, lbs (per belt or rib or ribs		(Eq. 3)
	ing that the tension created by belt tension and coefficient of the second second second second second second s	-	belt element is
$\frac{T_1}{T_2} = e^{\frac{f\phi}{\sin\beta}} = R$		(Reference 3)	
where:			
f = coefficient of fri			
$\varphi$ = the angle of con			
$\beta$ = "wedge" angle	of the V-belt		
R is used to determi	ne the tension ratio factor = $-\frac{1}{H}$	$\frac{R}{R-1}$	
$\frac{R}{R-1} = \frac{\frac{T_1}{T_2}}{\frac{T_1}{T_2} - 1} = \frac{T_1}{T_1 - 1}$	$\frac{T_1}{T_2} = \frac{T_1}{T_e}$		
Therefore, $T_{\rm I} = T_e \frac{R}{R-1}$ ,	$T_2 = \frac{T_1}{R} = T_e \frac{1}{R-1}$		
Substitute $T_1$ and $T_2$ into E	2		
Since f, $\varphi$ and $\beta$ are consta	ints, R is a constant, $\frac{T_{i1}}{T_{i2}} = \frac{T_{i1}}{T_{i2}}$	212	
Combine with Eq. 1 and E		**	(Eq. 4)
3.4 Required Pretens	ion		
	rand tension needs to be appli nsion or "static strand tension n:		-

$$T_{s} = \frac{63030 \cdot HP \cdot (2A_{R} - 1)}{N \cdot d \cdot N_{1}} + \frac{W \cdot d^{2}N_{1}^{2}}{1.69 \times 10^{6}}$$
 (Reference 2)

Where:

\_\_\_\_

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by interpolati $\frac{1.30 - 1.28}{0.30 - 0.20} =$ N = number of ribs d = smaller sheave N <sub>1</sub> = smaller sheave	factor = 1.2852 at $\frac{D-d}{C} = 0.2$ on of factors in Table 41 of R $\frac{x-1.28}{0.226-0.20}$ $x = \frac{0.0}{0.226-0.20}$ = 8 diameter, in = 15.0 in e speed, rpm = 900 rpm rib, lb/ft = 0.16 lb/ft	eference 2,	= A <sub>R</sub>
$T = \frac{63030 \times 100}{100}$	$\frac{2 \times 1.2852 - 1}{\times 900} + \frac{0.16 \times 15^2 \times 9}{1.69 \times 10^6}$	$\frac{00^2}{1080}$ = 1080 <i>lb</i>	

The required tension in the belt is  $T_S \cdot N = 108.9 \times 8 = 871.2 \text{ lb}$ 

# **3.5** Pretension Created by the Tensioning Procedure Specified in GM Maintenance Instruction (M.I. 1200)

It is specified on page 10 of General Motors Maintenance Instruction (M.I. 1200) (Reference 5) that, after installation of the belts on the sheaves, the belts need to be adjusted to the following condition. Each belt should deflect 7/16 in when the applied force is 10 to 13 pounds for each belt in a set of 8 belts. When the one 8-rib belt installed, the same procedure was followed. The average of 10 lb and 13 lb, or 11.5 lb, deflection force is used to calculate the belt pretension. The total deflection force on the set of 8 belts is assumed to be  $8 \times 11.5$  lb = 92 lb.

The deflection force for a V-belt is determined with the following equation:

$$F_L = \frac{T_s N}{16} + \frac{NKP}{L}$$
 (Reference 2) (Eq. 5)

where:

 $F_L$  = belt deflection force, lbs K = belt modulus factor P = span length, in L = belt length, in

Calculation No. 0083-0314-CZ $P = C \left[ 1 - 0.125 \left( \frac{D-d}{C} \right)^2 \right]$	Checked By (Reference 2)	Page: 8 Revision: 0
	(Reference 2)	Revision: 0
$P = C \left[ 1 - 0.125 \left( \frac{D - d}{C} \right)^2 \right]$	(Reference 2)	
$C = \frac{k + \sqrt{k^2 - 32 (D - d)^2}}{16}$	(Reference 4, p8-55)	(Eq. 6)
$k = 4 L_p - 6.28 (D + d)$	(Reference 4, p8-55)	(Eq. 7)
where: C = center distance between the two sheaves $L_p = belt pitch length, i.e. the effective length$	·	
$k = 4 \times 112 - 6.28 (21.2 + 15.0) = 220.66 in$		
$C = \frac{220.66 + \sqrt{220.66^2 - 32(21.2 - 15.0)^2}}{16} = 27.41$	1 in	
$P = 27.41 \left[ 1 - 0.125 \left( \frac{21.2 - 15.0}{27.41} \right)^2 \right] = 27.23 \text{ in}$		
From Eq. 5, $T_s N = 16 \left[ F_L - \frac{NKP}{L} \right]$		(Eq. 8)

Assume that the impact of misalignment of the upper and lower sheaves on the pretension is minimal. The assumed installation tension is:

$$T_s N = 16 \left[ 92 - \frac{8 \times 4.88 \times 27.23}{112} \right] = 1320 \text{ lbs}$$

Thus, the belt pretension of 1320 lbs exceeds the tension of 871.2 lbs, calculated in Section 3.4, required to transmit the engine drive horsepower to the fan shaft.

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3.6 Tension Chang	e from Pillow Block Mover	nent	
Initial Belt Length Calc	ulation:		
By Hooke's Law,	$F = k \cdot \Delta L / L_i$		
Where: k = the belt modul $\Delta L =$ the belt leng	us = 160,000 lb per 100% elon th change	gation per belt or rib (Re	ference 10)
Under the belt pretensio	n of $F = 1320$ lbs,		
$\Delta L = F \cdot L_i / k = 13$	20 x 112 / (160,000 x 8) = 0.11	55 in	
The initial belt length Li	$= L + \Delta L = 112.1155$ in		
Minimum Center Distan	ce Calculation:		
	OYSTER CREEK		
	Pedestal Pil	= <sup>C</sup> i = 27.41 in low Block hal Peak Position Τ Β	
Figure 1. Illustration of	of the Belt Drive Misalignmen Position	t and the Pillow Block V	ibration Peak

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block position after the found on EDG 1 at Oys	C is the pillow block original in lower bearing bolt was lost and t ter Creek. T'B is the pillow bloc the upper and lower sheaves is t	he upper bolt backed out k vibration peak position	1/2 inch as
After the belt is pretensi	oned, the center distance is calcu	ulated with Eq. 6 and Eq.	.7:
$k = 4 \times 112.1155$	- 6.28 (21.2 + 15.0) = 221.126 ir	1	
$C_i = \frac{221.126 + \sqrt{2}}{2}$	$\frac{221.126^2 - 32(21.2 - 15.0)^2}{16} = 2$	7.466 in	
By scaling the distance f M.I. 1200 (Reference 5)	from the pillow block top edge to $TC = 4.95$ in.	the bearing center line of	on page 3 of
From Reference 8, $A_1B$	$= AA' - AA_1 + A'B = \frac{1}{2} - \frac{1}{4} + \frac{3}{4} = 1$	1.00 in	
	$\frac{2}{2-A'B^2} = \sqrt{TC^2 - A'B^2} = \sqrt{4.9}$		
$OT_1 \approx OC - TC =$	$C_i - TC = 27.466 - 4.95 = 22.516$	5 in	
$OA_1 = OT_1 + T_1A_1 =$	= 22.516 + 4.89 = 27.406 in		
$C_f = OB = \sqrt{OA_1^2}$	$\overline{A_1B^2} = \sqrt{27.406^2 + 1.00^2} = 27.406^2 + 1.00^2$	424 in	
Belt Length Relaxation			
The belt length is calculated	ated with:		
$L = 2 \cdot C \cdot \cos \theta + \pi$	$(D + d)/2 + \pi \cdot \theta (D - d)/180$	(Reference 6)	
Where: $\theta = \sin^{-1}(\frac{D-d}{2C})$	)		
The initial belt length:			
$\theta = \sin^{-1}(\frac{D-d}{2C}) =$	$=\sin^{-1}(\frac{21.2-15.0}{2\times27.424})=6.4906^{\circ}$		

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Tension Loss Because of Belt Length Relaxation

 $\Delta T_s N = 8 \times 160,000 \times 0.0539/112.1155 = 615.0 \text{ lbs}$ 

#### 3.7 Calculation of Fan Speed on MP45A

From Eq. 5.3 of Reference 11, the fan horsepower varies as the cube of the shaft speed:

$$\frac{P_1}{P_2} = \left(\frac{N_1}{N_2}\right)^3$$
(Eq. 9)
where:  $\mathbf{P} = \text{fan horsenower} = \pi \cdot \mathbf{e} = \pi \cdot 2\pi \mathbf{N}$ 

where: P = fan horsepower  $= \tau \cdot \omega = \tau \cdot 2\pi N$ 

Substitute into Eq. 9 and reorganize:  $\frac{\tau_1}{\tau_2} = \frac{N_1^2}{N_2^2}$ 

Combine with Eq.4:  $\frac{(T_s N)_1}{(T_s N)_2} = \frac{N_1^2}{N_2^2}$ 

The belt tension in the fan belt drive:  $(T_sN)_1 = 1320 - 615 = 705$  lb

The required belt tension:  $(T_sN)_2 = 871.2$  lb

$$\frac{N_1^2}{N_2^2} = \frac{705.0}{871.2} = 80.92\% \qquad \qquad \frac{N_1}{N_2} = \sqrt{80.92\%} = 89.95\%$$

The fan shaft speed with belt slippage on MP45A is

 $N_1 = 89.95 \% \times 637 \ rpm = 573 \ rpm$ 

The required tension due to the centrifugal force of the belt element is by Reference 2

 $\frac{Wd^2(rpm)^2}{1.69 \times 10^6} = \frac{0.16 \times 21.2^2 (573)^2}{1.69 \times 10^6} = 13.97 \text{ lb}, \text{ which is small enough to be omitted. This}$ 

confirms the assumption made on page 4.

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#### 3.8 Calculation of Air Flow at Lowered Fan Speed

From Eq. 5.1 of Reference 11,  $\frac{Q_1}{Q_2} = \frac{N_1}{N_2}$ 

From Reference 8, the air flow at rated speed (637 rpm) is 125,000 cfm. The air flow at lowered fan speed of 573 rpm, the air flow is

$$Q_1 = \frac{N_1}{N_2}Q_2 = \frac{573}{637} \times 125000 = 112,400 \text{ cfm}$$

#### 3.9 Belt Life Estimation

As calculated in Section 3.7, the belt tension decreased to 80.92% of required tension. From tension vs belt life curve (Reference 6) from Goodyear, at 20% decrease in tension, the belt life decrease percentage is:

80% + 31/32 x 20% = 99.375%

The belt life is estimated as (1 - 99.375%) (Performance Index) = 618.75 hr

Where the performance index = belt life = 99,000 hr (Reference 1)

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4.0 REFEREN	CES		· • • • • • • • • • • • • • • • • • • •
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	e and T. Baumeister III, Mark's tion. 1996 The McGraw-Hill Co		lechanical
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7. Email to Exelon (M @8:53am.	r. Arvin Ho) from ESI (Robin V	Veeks) dated on August 16	5, 2004
	Creek EDG Cooling Fan Drive l dated August 18, 2004.	e Test-Technical Backgrou	ind and Basis",
	f an EMD MP Radiator Fan Dri <sup>.</sup> Bolts" Rev. 0, August 10, 2004.	ve with Degraded Lower I	Pillow Block
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	CALCULATION T	ITLE PAGE		
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Title: EDG Heatup With Redu	uced Radiator Heat Removal (	Capability	Calculation No. 0083-0314-RCS01	
Preparer / Date	Checker / Date	Reviewer & Approver	/ Date	Rev. No.
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		RECORD OF REV	ISIONS	
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Revision	Affected Pages	Description		
0	All	Initial Issue		
1	All		rate to be consistent with for 90 °F ambient air temp rates to heatup curves for	erature.

**Note:** The revision number found on each individual page of the calculation carries the revision level of the calculation in effect at the time that page was last revised.

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Appendix D Cooling S	ystem Heat Absorbing C		

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#### PURPOSE

The purpose of this calculation is to determine the heatup rate for the Oyster Creek EDG with reduced radiator heat removal capability.

#### BACKGROUND

Oyster Creek recently experienced an incident in which the drive pulley for the EDG radiator fan became loose, resulting in a loss of tension in the fan drive belts and reduced air flow through the radiator. This calculation determines the effect of reduced radiator air flow on the heatup rate of the EDG.

#### RESULTS

The results of this calculation are shown in Figures 2 and 3.

Figure 2 shows the temperature of the coolant entering the engine assuming the EDG is operating at its rated power of 2600 KWe with an ambient air temperature of 70 °F. Coolant inlet temperature is shown for air flow rates ranging from design flow (125,000 scfm) to 55% of design flow (68,750 scfm).

Figure 3 shows the temperature of the lube oil entering the engine assuming the EDG is operating at its rated power of 2600 KWe, with an ambient air temperature of 70 °F. Lube oil inlet temperature is shown for air flow rates ranging from design flow (125,000 scfm) to 55% of design flow (68,750 scfm).

Also shown in Figures 2 and 3 are temperatures measured on Oyster Creek EDG 1 on 4/30/04 (Reference 10), when the ambient air temperature was about 70 °F and the EDG was operating at about 2700 KWe. These data show that the analytical model developed in this calculation agrees well with measured operating data.

#### ASSUMPTIONS

1. Radiant and convective heat losses from the surfaces of the EDG and associated components are neglected. All of the heat removed from the EDG is through the radiator. This is conservative since radiant and convective heat losses will reduce the heatup rate.

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2. T	he heat added by	the cooling water pump and	lube oil pump is neglig	ible.
		erties of the coolant are const emperature on the coolant pl	-	
		rties of the lube oil are const rature on the lube oil physic		
ri	se of the coolant i	nperature rise of the coolant in the engine is constant and ne heat added in the engine a	equal to the ratio of the	heat added in the
	ne heat absorbing egligible.	capacity [(mass)X(specific ]	heat)] of the air in the ra	adiator is
	ne efficiency of th f 0.40	ne fins in the radiator remain	s constant at the calcula	ted design value
	ne efficiency of that alue of 0.93	ne fins in the lube oil cooler i	remains constant at the	calculated design
	ne initial temperat mperature.	tures of the coolant and lube	oil are equal to the amb	bient air
of ra si	the radiator for t diator for the Oys zes of the engines	at removal capability [(heat t he test EDG (Model MP-36) ster Creek EDG (Model MP- s (16 cylinders compared to 2 lytical heatup model.	to the heat removal cap 45) is equal to the ratio	oability of the of the physical
	r the test EDG (N	tt absorbing capacity [(mass) Model MP-36) to the heat abs		cooling system for

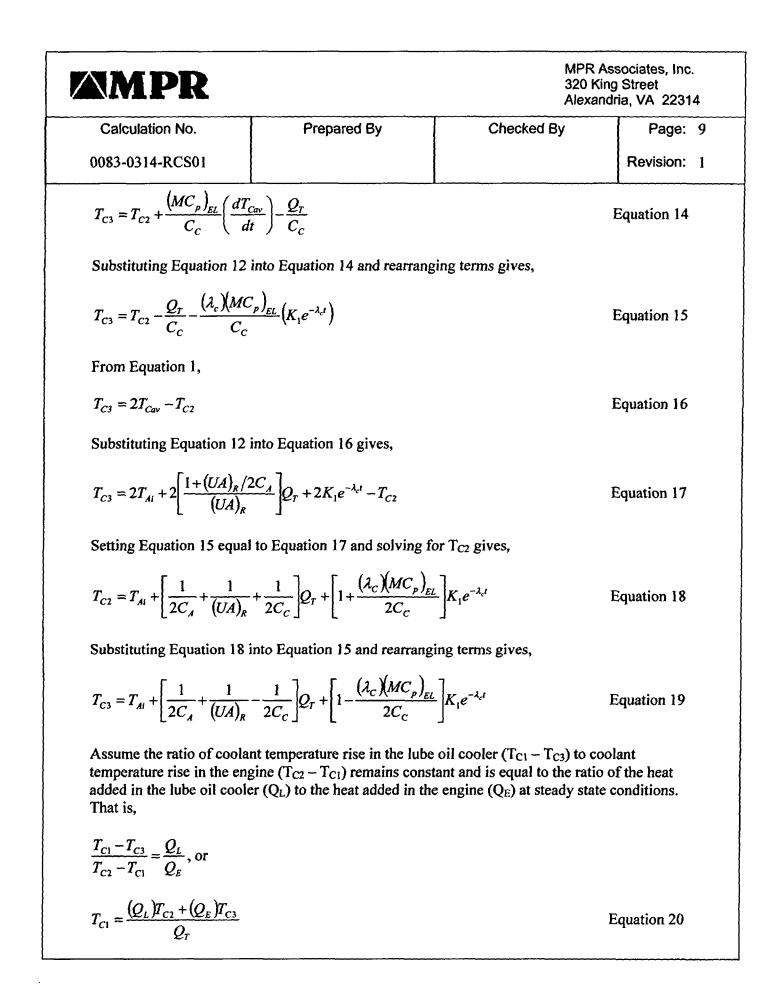
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Analysis			
The heatup rates for the	of the Oyster Creek EDG radiato temperature of the cooling water oil entering the engine $(T_{L1})$ are o	entering the engine (T <sub>C</sub>	) and the
Cooling Water Heatu	ıp Rate		
Define,			
$T_{Cav} = \frac{1}{2} (T_{C2} + T_{C3})$			Equation 1
$T_{Aav} = \frac{1}{2} \left( T_{Ai} + T_{Ao} \right)$			Equation 2
Where,			
$T_{Cav}$ = the average coola	nt temperature in the cooling sys	tem, °F	
$T_{C2}$ = the coolant temper	rature exiting the engine (entering	g the radiator), °F	
$T_{C3}$ = the coolant temper	rature entering the lube oil cooler	(exiting the radiator), °I	7
$T_{Aav}$ = the average air terms	mperature in the radiator, °F		
$T_{Ai}$ = the air temperature	entering the radiator, °F		
$T_{Ao}$ = the air temperature	e exiting the radiator, °F		
A transient heat balance equation,	on the air in the radiator can be a	pproximated with the fo	llowing
$\left(MC_{p}\right)_{A}\left(\frac{dT_{Aav}}{dt}\right) = \left(UA\right)_{R}$	$(T_{Cav} - T_{Aav}) - C_A (T_{Ao} - T_{Ai})$		Equation 3

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$(MC_p)_A$ = heat absorbing	capacity [(mass)X(specific he	at)] of the air in the radiat	tor, Btu/°F
(UA) <sub>R</sub> = effective produc radiator, Btu/hr	t of heat transfer coefficient ti °F	nes heat transfer area for	the
C <sub>A</sub> = heat transport capac the radiator, Btu/hr	ity [(mass flow rate)X(specifients)	c heat)] of air flowing thro	ough
Assuming that $(MC_p)_A$ is	negligible, Equation 3 become	?S	
$0 = (UA)_R (T_{Cav} - T_{Aav}) - C$	$T_A(T_{Ao}-T_{Ai})$ , or		
$T_{Ao} = T_{Ai} + [(UA)_R / C_A] (T_C)$	$T_{av} - T_{Aav}$		Equation 4
From Equation 2,			
$T_{Ao} = 2T_{Aav} - T_{Ai}$			Equation 5
Setting Equation 4 equal	to Equation 5 and solving for 7	T <sub>Aav</sub> gives,	
$T_{Aav} = \frac{T_{Ai} + [(UA)_R/2C_A]}{1 + [(UA)_R/2C_A]}$	<u>Cav</u>		Equation 6
	nvective heat losses from the one and lube oil cooler can be a		
$\left(MC_{p}\right)_{EL}\left(\frac{dT_{Cav}}{dt}\right) = Q_{T} - Q_{T}$	$C_c(T_{C2}-T_{C3})$		Equation 7
Where,			
	absorbing capacity [(mass)X(s be oil cooler (including associa		nt in the
$Q_T$ = total heat added to the total heat added total heat add	ne coolant (engine plus lube oi	l cooler), Btu/hr	
$C_C$ = heat transport capacities the coolant system, 2	ity [(mass flow rate)X(specific Btu/br °F	heat)] of coolant flowing	through

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A transient heat balance of equation,	on the coolant in the radiator ca	n be approximated with	the following
$\left(MC_p\right)_R\left(\frac{dT_{Cav}}{dt}\right) = C_C\left(T_{Cav}\right)$	$(T_{C3}) - (UA)_R (T_{Cav} - T_{Aav})$		Equation 8
Where,			
	bsorbing capacity [(mass)X(spo ing associated metal), Btu/°F	ecific heat)] of the coola	nt in the
Adding Equations 7 and 8	3 gives,		
$\left(MC_{p}\right)_{CT}\left(\frac{dT_{Cav}}{dt}\right) = Q_{T} - \left(\frac{dT_{Cav}}{dt}\right)$	$UA)_{R}(T_{Cav}-T_{Aav})$		Equation 9
Where,			
$\left(MC_{p}\right)_{CT}=\left(MC_{p}\right)_{EL}+\left(M\right)_{EL}$	$C_p \Big)_R$		Equation 10
Substituting Equation 6 in	to Equation 9 and rearranging	terms gives,	
$\left(MC_{p}\right)_{CT}\left(\frac{dT_{Cav}}{dt}\right) + \left[\frac{dT_{Cav}}{1+(U)}\right]$	$\frac{UA)_R}{UA)_R/2C_A} \bigg] T_{Cav} = Q_T + \bigg[ \frac{(UA)_R}{1 + (UA)_R} \bigg]$	$\frac{)_{R}}{R/2C_{A}} \bigg] T_{Ai}$	Equation 11
The solution to Equation	11 is		
$T_{Cav} = T_{Ai} + \left[\frac{1 + (UA)_R/2C}{(UA)_R}\right]$	$\left[ \frac{\Delta}{\Delta} \right] Q_T + K_1 e^{-\lambda_c t}$		Equation 12
Where,			
$K_1 = constant of integration$	on		
$\lambda_c = \frac{(UA)_R}{(MC_R)_{cr} [1 + (UA)_R/2]}$			Equation 13

From Equation 7,



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Where,			
$Q_T = Q_E + Q_L$			Equation 21
Substituting Equations 18	8 and 19 into Equation 20 and	rearranging terms gives,	
$T_{C1} = T_{Ai} + \left[\frac{1}{2C_A} + \frac{1}{(UA)_R}\right]$	$-\frac{(Q_E - Q_L)}{2C_C Q_T} \bigg] Q_T + \bigg[ 1 - \frac{(Q_E - Q_L)}{2C_C Q_T} \bigg] Q_T + $	$\frac{Q_L(\lambda_C)(MC_p)_{EL}}{2C_CQ_T}\bigg]K_1e^{-\lambda_c t}$	Equation 22
As t goes to infinity, $T_{C1}$	goes to T <sub>C1f</sub> , where		
$T_{C1f}$ = the final steady sta	te value of T <sub>C1</sub> , °F		
$T_{C1f} = T_{Ai} + \left[\frac{1}{2C_A} + \frac{1}{(UA)}\right]$	$\frac{-(Q_E - Q_L)}{2C_C Q_T} \bigg] Q_T$		Equation 23
Therefore,			
$T_{C1} = T_{C1f} + \left[1 - \frac{(Q_E - Q_L)}{2}\right]$	$\frac{(\lambda_c)(MC_p)_{EL}}{C_c Q_T} \bigg] K_1 e^{-\lambda_c t}$		Equation 24
At t equals zero, T <sub>C1</sub> equa	ls T <sub>C1i</sub> , where		
$T_{C1i}$ = the initial value of	T <sub>C1</sub> , °F		
Therefore,			
$T_{C1f} + \left[1 - \frac{(\mathcal{Q}_E - \mathcal{Q}_L)(\lambda_C)}{2C_C \mathcal{Q}_T}\right]$	$\frac{MC_p\big)_{EL}}{K_1} = T_{C11}, \text{ or }$		
$K_1 = \frac{-2C_c Q_r (T_{CIF})}{2C_c Q_T - (Q_E - Q_L)}$	$\frac{-T_{Cli}}{\lambda_{C}} \left( MC_{p} \right)_{EL}$		Equation 25
Therefore,			

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$T_{C1} = T_{C1f} - (T_{C1f} - T_{C1i})e^{-T_{C1i}}$	λ <sub>c</sub> t		Equation 26	
From Appendices A throu	ıgh D,			
$Q_E = 5.123X10^6 Btu / hr$				
$Q_L = 1.986X10^6 Btu / hr$				
$Q_T = 7.109 X 10^6 Btu / hr$				
$C_c = 5.394 X 10^5 Btu / hr$	°F			
$C_L = 8.402 X 10^4 Btu / hr$	°F			
$C_A = 1.073G_A Btu/hr \circ F$	, $G_A$ in scfm			
$(UA)_{R} = \frac{1.838 \times 10^{5}}{\left[1 + \frac{1426.22}{G_{A}^{0.60}}\right]} Bt$	u/hr °F , G <sub>A</sub> in scfm			
$(UA)_{C} \approx 6.729 X 10^{4} Btu/h$	r °F			
$\left(MC_p\right)_{CT} = 1.182X10^4 Bt$	u /° F			
	with Equations 13, 23 and 26, ates with an ambient air tempe			

## Lube Oil Heatup Rate.

Define,

$$T_{CLav} = \frac{1}{2} (T_{C1} + T_{C3})$$
$$T_{Lav} = \frac{1}{2} (T_{L1} + T_{L2})$$

Equation 27

Equation 28

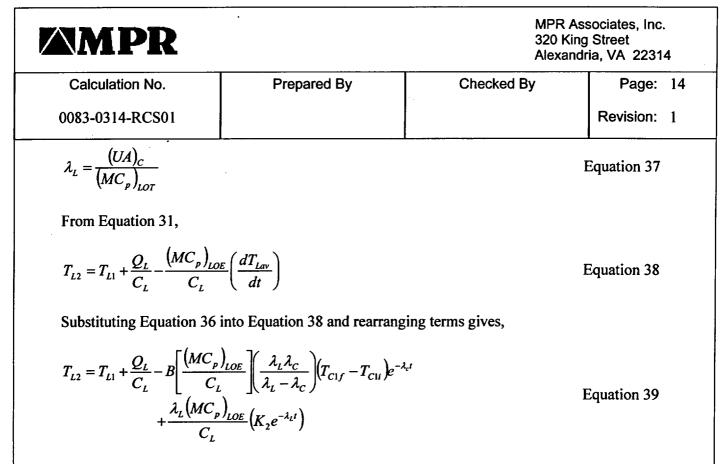
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Where,				
$T_{CLav}$ = the average coola	nt temperature in the lube oil	cooler, °F		
$T_{C1}$ = the coolant temperature	ature exiting the lube oil coole	r (entering the engine), °F	7	
$T_{C3}$ = the coolant temperature	ature entering the lube oil cool	er (exiting the radiator), °	F	
$T_{Lav}$ = the average lube o	il temperature, °F			
$T_{L1}$ = the lube oil temperature	ature entering the engine (exit	ng the lube oil cooler), °F	·	
$T_{L2}$ = the lube oil temperature	ature exiting the engine (enter	ng the lube oil cooler), °F		
Substituting Equations 19 and 25 gives,	and 22 into Equation 27, rear	ranging terms and using H	Equations 23	
$T_{CLav} = T_{Clf} - \left(\frac{Q_L}{2C_C}\right) - B$	$(T_{C1f} - T_{C1i})e^{-\lambda_c t}$		Equation 29	
Where,				
$B = \frac{2C_c Q_T - Q_E \lambda_c (M)}{2C_c Q_T - (Q_E - Q_L) \lambda_c}$	$\frac{C_p_{eL}}{\left(MC_p_{eL}\right)_{EL}}$		Equation 30	
A transient heat balance of equation,	on the lube oil in the engine ca	n be approximated with th	e following	
$\left(MC_{p}\right)_{LOE}\left(\frac{dT_{Lav}}{dt}\right) = Q_{L} - $	$C_L(T_{L2}-T_{L1})$		Equation 31	
Where,				
	absorbing capacity [(mass)X( ding associated metal), Btu/°F		oil in the	

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C<sub>L</sub> = heat transport capacity [(mass flow rate)X(specific heat)] of lube oil flowing through the lube oil system, Btu/hr °F

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A transient heat balance of following equation,	n the lube oil in the lube oil co	oler can be approximated	l with the
$\left(MC_{p}\right)_{LOC}\left(\frac{dT_{Lav}}{dt}\right) = C_{L}\left(T_{Lav}\right)$	$(L_2 - T_{L1}) - (UA)_C (T_{Lav} - T_{CLav})$		Equation 32
Where,			
	absorbing capacity [(mass)X(s r (including associated metal),		oil in the
(UA) <sub>C</sub> = effective product lube oil cooler, B	of heat transfer coefficient tim Btu/hr °F	es heat transfer area for	the
Adding Equations 31 and	32 gives,		
$\left(MC_{p}\right)_{LOT}\left(\frac{dT_{Lav}}{dt}\right) = Q_{L} - Q_{L}$	$(UA)_{C}(T_{Lav}-T_{CLav})$		Equation 33
Where,			
$\left(MC_{p}\right)_{LOT}=\left(MC_{p}\right)_{LOE}+\left(MC$	$MC_p \Big _{LOC}$		Equation 34
Substituting Equation 29 i	nto Equation 33 and rearrangin	g terms gives,	
$\left(MC_{p}\right)_{LOT}\left(\frac{dT_{Lav}}{dt}\right) + \left(UA\right)_{C}$	$T_{Lav} = (UA)_{C} T_{C1f} + (UA)_{C} \left[ \frac{1}{(UA)_{C}} - B(UA)_{C} (T_{C1f} - T_{C1f}) \right]$		Equation 35
The solution to Equation 3	5 is		
$T_{Lav} = T_{C1f} + \left[\frac{1}{(UA)_C} - \frac{1}{2C_C}\right]$	$\frac{1}{c} \left[ Q_L - B\left(\frac{\lambda_L}{\lambda_L - \lambda_C}\right) \left(T_{C1f} - T_{C1f}\right) \right] \right]$	$e^{-\lambda_c t} + K_2 e^{-\lambda_L t}$	Equation 36
Where,			

•



From Equation 28,

$$T_{L2} = 2T_{Lav} - T_{L1}$$
 Equation 40

Substituting Equation 36 into Equation 40 gives,

$$T_{L2} = 2T_{C1f} + 2\left[\frac{1}{(UA)_{C}} - \frac{1}{2C_{C}}\right]Q_{L} - 2B\left(\frac{\lambda_{L}}{\lambda_{L} - \lambda_{C}}\right)(T_{C1f} - T_{C1i})e^{-\lambda_{c}t} + 2K_{2}e^{-\lambda_{L}t} - T_{L1} \quad \text{Equation 41}$$

Setting Equation 39 equal to Equation 41 and solving for  $T_{L1}$  gives,

$$T_{L1} = T_{C1f} + \left[\frac{1}{(UA)_{C}} - \frac{1}{2C_{C}} - \frac{1}{2C_{L}}\right]Q_{L}$$
  
$$-B\left(\frac{\lambda_{L}}{\lambda_{L} - \lambda_{C}}\right)\left[1 - \frac{\lambda_{C}(MC_{p})_{LOE}}{2C_{L}}\right](T_{C1f} - T_{C1i})e^{-\lambda_{c}t} + K_{2}\left[1 - \frac{\lambda_{L}(MC_{p})_{LOE}}{2C_{L}}\right]e^{-\lambda_{L}t}$$
 Equation 42

As t goes to infinity,  $T_{L1}$  goes to  $T_{L1f}$ , where

 $T_{L1f}$  = the final steady state value of  $T_{L1}$ , °F

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$T_{L1f} = T_{C1f} + \left[\frac{1}{(UA)_C} - \frac{1}{2C}\right]$	$\frac{1}{C_c} - \frac{1}{2C_L} \bigg] Q_L$		Equation 43

Therefore,

$$T_{L1} = T_{L1f} - B\left(\frac{\lambda_L}{\lambda_L - \lambda_C}\right) \left[1 - \frac{\lambda_C \left(MC_p\right)_{LOE}}{2C_L}\right] \left(T_{C1f} - T_{C1i}\right) e^{-\lambda_c t} + K_2 \left[1 - \frac{\lambda_L \left(MC_p\right)_{LOE}}{2C_L}\right] e^{-\lambda_L t}$$
Equation 44

At t equals zero,  $T_{L1}$  equals  $T_{L1i}$ , where

 $T_{L1i}$  = the initial value of  $T_{L1}$ , °F

Therefore,

$$T_{L1f} - B\left(\frac{\lambda_L}{\lambda_L - \lambda_C}\right) \left[1 - \frac{\lambda_C \left(MC_p\right)_{LOE}}{2C_L}\right] \left(T_{C1f} - T_{C1i}\right) + K_2 \left[1 - \frac{\lambda_L \left(MC_p\right)_{LOE}}{2C_L}\right] = T_{L1i}, \text{ or}$$

$$K_2 = \frac{B\left(\frac{\lambda_L}{\lambda_L - \lambda_C}\right) \left[2C_L - \lambda_C \left(MC_p\right)_{LOE}\right] \left(T_{C1f} - T_{C1i}\right) - 2C_L \left(T_{L1f} - T_{L1i}\right)}{2C_L - \lambda_L \left(MC_p\right)_{LOE}}$$
Equation 45

Substituting Equation 45 into Equation 44 and rearranging terms gives,

$$T_{L1} = T_{L1f} + \left(\frac{B}{2C_L}\right) \left(\frac{\lambda_L}{\lambda_L - \lambda_C}\right) \left[2C_L - \lambda_c (MC_P)_{LOE}\right] \left(T_{C1f} - T_{C1i}\right) \left[e^{-\lambda_L t} - e^{-\lambda_c t}\right]$$
  
=  $-\left(T_{L1f} - T_{L1i}\right) e^{-\lambda_L t}$  Equation 46

In Appendix D, it is shown that  $\lambda_L$  is approximately equal to  $\lambda_C$ . In this case, Equation 46 becomes,

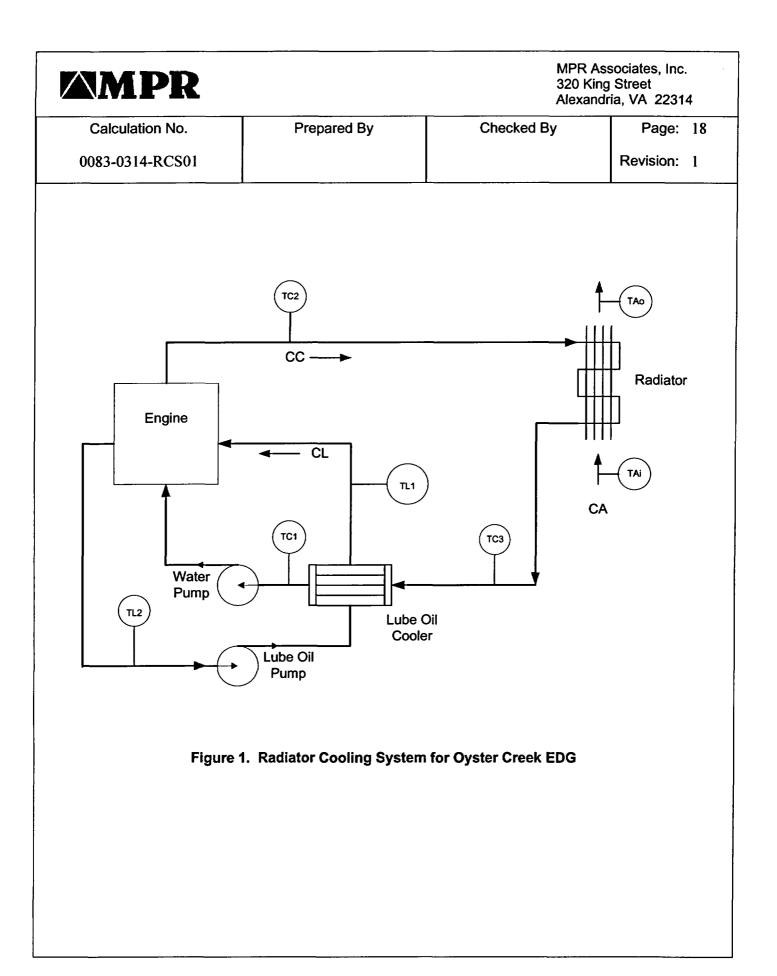
$$T_{L1} = T_{L1f} - \left(T_{L1f} - T_{L1i}\right)e^{-\lambda_c t}$$
 Equation 47

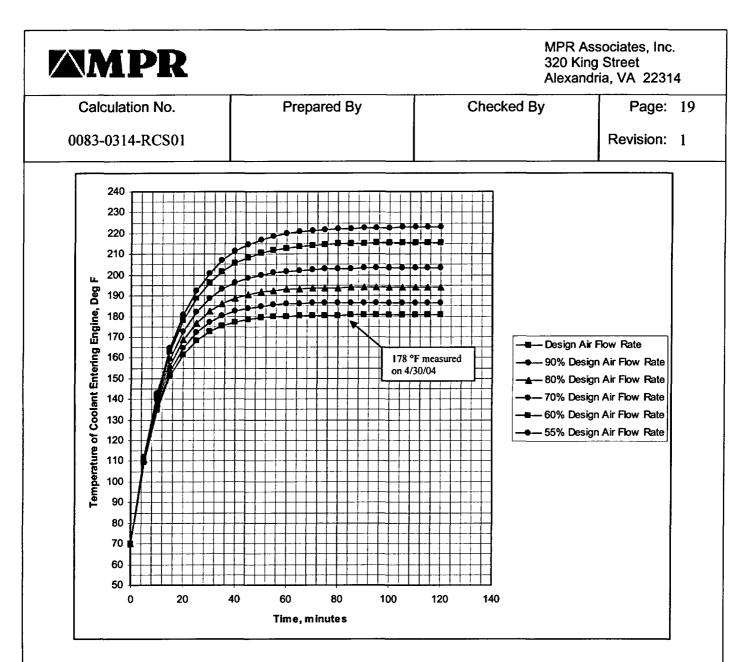
From Appendices A through D,

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$Q_E = 5.123X10^6 Btu / hr$			
$Q_L = 1.986 X 10^6 Btu / hr$			
$Q_T = 7.109 X 10^6 Btu / hr$			
$C_c = 5.394 X 10^5 Btu / hr °F$			
$C_L = 8.402 X 10^4 Btu / hr °F$			
$C_A = 1.073G_A Btu/hr^{\circ}F$ , G	a in scfm		
$(UA)_{R} = \frac{1.838X10^{5}}{\left[1 + \frac{1426.22}{G_{A}^{0.60}}\right]} Btu/H$	$F  ^{\circ}F$ , $G_{A}$ in scfm		
(UA) <sub>C</sub> = 6.729X10 <sup>4</sup> Btu/hr °F	,		
$\left(MC_{p}\right)_{CT}=1.182X10^{4} Btu/^{\circ}$	F		

Using these values along with Equations 13, 23, 26, 43 and 47,  $T_{L1}$  has been calculated as a function of time for several air flow rates with an ambient air temperature of 70 °F. The results are shown in Figure 3.

Calculation No.	Prepared By	Checked By	andria, VA 22314 Page: 1
	T Tepared by		
0083-0314-RCS01			Revision: 1
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Nuclear (A. Ho),	Telephone Conversation betw Subject: EMD Model 645 Die s, dated August 25, 2004.	•	d Exelon
Diesel Power Uni	ower Application and Installa ts, 8-12-16 Cylinder Roots Bl strial Applications," General M	own, 8-12-16-20 Cylinder	Turbocharged,
4. F. Kreith and M. F Company, 1997.	Bohn, "Principles of Heat Tra	nsfer," Fifth Edition, PWS	Publishing
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7. General Motors E Assembly (Oil Co	lectro-Motive Division Drawi oler)."	ng 9514842, Rev B, "Core	e & Header
With Degraded Lo	r: 6012458-TP-1, Rev 0, "Tes ower Pillow Block Bearing M lear Generating Station, July	ounting Bolts," Exelon Ar	
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	gen Energy (D. Jones) to MPI		EDG1
	meters, dated August 26, 2004		

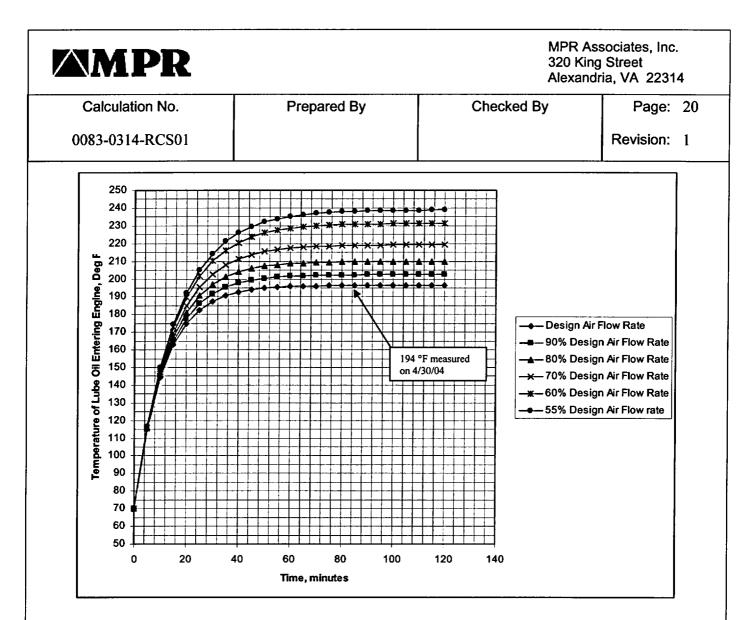




Design Air Flow Rate = 125,000 scfm

Steady State Coolant Outlet Temperature = Steady State Coolant Inlet Temperature + 9.5 °F

#### Figure 2. Temperature of Cooling Water Entering Oyster Creek EDG Engine 70°F Ambient Air Temperature



Design Air Flow Rate = 125,000 scfm

Steady State Lube Oil Outlet Temperature = Steady State Lube Oil Inlet Temperature + 23.6 °F

#### Figure 3. Temperature of Lube Oil Entering Oyster Creek EDG Engine 70°F Ambient Air Temperature

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#### PURPOSE

The purpose of this appendix is to determine the heat loads  $(Q_E, Q_L, Q_T)$  for the Oyster Creek EDG cooling system. The references used in this appendix are listed in the Reference section of the main body of this calculation.

#### ANALYSIS

From References 1 and 2, the Oyster Creek EDGs are General Motors Model MP-45 units with 20 cylinder diesel engines rated at 3485 bhp. From Reference 2, the heat rejected to the jacket water is 24.5 Btu/min per bhp and the heat rejected to the lube oil is 9.5 Btu/min per bhp. Therefore,

$Q_E = \left(\frac{24.5 Btu}{\min bhp}\right) \left(\frac{60 \min}{hr}\right) (3485 bhp)$	
$Q_E = 5.123X10^6 Btu / hr$	Equation A-1
$Q_L = \left(\frac{9.5 Btu}{\min bhp}\right) \left(\frac{60 \min}{hr}\right) (3485 bhp)$	
$Q_L = 1.986X10^6 Btu/hr$	Equation A-2
$Q_T = Q_E + Q_L = 5.123X10^6 + 1.986X10^6$	
$Q_T = 7.109X10^6 Btu / hr$	Equation A-3

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# **Heat Transport Capacities**

## PURPOSE

The purpose of this appendix is to determine the heat transport capacities,  $C_C$ ,  $C_L$ ,  $C_A$ , [(mass flow rate)X(specific heat)] for the fluids (cooling water, lube oil, air) used in the Oyster Creek EDG cooling system. The references used in this appendix are listed in the Reference section of the main body of this calculation.

## ANALYSIS

### Cc

From page 2-4 of Reference 3, the engine cooling water flow rate is 1100 gpm. From Reference 1, Oyster Creek does not use anti-freeze; therefore, this analysis will use the physical properties of fresh water. From Reference 2, the radiators were sized for a maximum engine water outlet temperature of 210 °F with an ambient temperature of 90 °F. Therefore, assume an average temperature of about 150 °F. From Table 13 (page A14) of Reference 4,

$$\rho_{\rm C} = 61.2 \ \rm lb/ft^3$$

 $(C_p)_C = 0.999 \text{ Btu/lb }^\circ\text{F}$ 

$$C_c = \left(\frac{1100 \text{ gal}}{\min}\right) \left(\frac{60 \min}{hr}\right) \left(\frac{1 \text{ ft}^3}{7.4805 \text{ gal}}\right) \left(\frac{61.2 \text{ lb}}{\text{ft}^3}\right) \left(\frac{0.999 \text{ Btu}}{\text{lb}^\circ F}\right)$$

$$C_{c} = 5.394 X 10^{5} Btu / hr ^{\circ}F$$

Equation B-1

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pump flow rate). From Re oil outlet temperature of 2	FReference 3, the engine lube of eference 2, the lube oil coolers 40 °F with an ambient tempera out 165 °F. From Table 16 (pa	were sized for a maximu ture of 90 °F. Therefore	im engine lube , assume an	
$\rho_L = 53.4 \text{ lb/ft}^3$				
$(C_p)_L = 0.503 \text{ Btu/lb }^{\circ}\text{F}$				
$C_L = \left(\frac{390 \ gal}{\min}\right) \left(\frac{60 \ \min}{hr}\right)$	$\left(\frac{1 ft^3}{7.4805 gal}\right) \left(\frac{53.4 lb}{ft^3}\right) \left(\frac{0.503}{lb^3}\right)$	$\left(\frac{Btu}{F}\right)$		
$C_L = 8.402 X 10^4 Btu / hr^2$	F		Equation B-2	2
C <sub>A</sub>				
	ign air flow rate is 125,000 scfi onditions (14.7 psia, 60 °F),	n. From Table 27 (page	A26) of	
$\rho_{A} = 0.0739 \text{ lb/ft}^{3}$				
$(C_p)_A = 0.242 \text{ Btu/lb }^\circ\text{F}$				
$C_{A} = \left(\frac{G_{A} ft^{3}}{\min}\right) \left(\frac{60 \min}{hr}\right) $	$\frac{0.0739 \ lb}{ft^3} \left( \frac{0.242 \ Btu}{lb \ F} \right)$			
$C_A = 1.073G_A Btu/hr^{\circ}F$	G <sub>A</sub> in scfm		Equation B-3	
$C_{A} = 1.341 X 10^{5} Btu / hr^{\circ}$	F, at design flow			

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C Heat Excha	inger Characteris	stics	

### PURPOSE

The purpose of this appendix is to determine the heat exchanger characteristics  $[(UA)_R, (UA)_C]$ for the radiator and lube oil cooler used in the Oyster Creek EDG cooling system. The references used in this appendix are listed in the Reference section of the main body of this calculation.

### ANALYSIS

#### $(UA)_R$

The radiator design information used in the following analysis is from Reference 5.

The radiator is a finned tube cross flow heat exchanger with the cooling water flowing on the insides of the tubes and the air flowing over the finned outside surfaces of the tubes. From page 124 of Reference 6, the overall heat transfer coefficient for the radiator is given by

$$U = \left[\frac{1}{E_f h_o} + \frac{r_o}{E_f} + r_w + \left(\frac{A_o}{A_i}\right)\left(\frac{1}{h_i} + r_i\right)\right]^{-1}$$
Equation

Where,

U = the overall heat transfer coefficient, Btu/hr  $ft^2 \circ F$ 

 $h_0 = film$  heat transfer coefficient on outside surfaces of tubes, Btu/hr ft<sup>2</sup> °F

 $h_i$  = film heat transfer coefficient on inside surfaces of tubes. Btu/hr ft<sup>2</sup> °F

 $r_o$  = thermal fouling resistance on outside surfaces of tubes, hr ft<sup>2</sup> °F/Btu

n C-1

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$r_i$ = thermal fouling resist	ance on inside surfaces of tube	es, hr ft <sup>2</sup> °F/Btu	
$r_w =$ thermal resistance of	tube walls, hr ft <sup>2</sup> °F/Btu		
$E_f = fin efficiency$			
$A_o = total outside heat tra$	nsfer surface area, ft <sup>2</sup>		
$A_i = total inside heat trans$	sfer surface area, ft <sup>2</sup>		
From pages 286 and 290	of Reference 6,		
$r_o = 0.001$ hr ft <sup>2</sup> °F/Btu, c	ompressed air		
$r_i = 0.001$ hr ft <sup>2</sup> °F/Btu, er	gine jacket water		
	red brass tubes with an OD of f each tube is 0.450 inch (0.50 f the tube walls is,		
$r_{w} = \left(\frac{OD}{24k_{t}}\right) \ln\left(\frac{OD}{ID}\right)$			
Where,			
$k_t = thermal conductivity$	of the tube material, Btu/hr ft	ŶF	
From Table 10 (page A9)	of Reference 4.		

 $k_t = 35.2 \text{ Btu/hr ft }^\circ\text{F}, \text{ red brass}$ 

Therefore,

$$r_{w} = \left(\frac{0.500}{24(35.2)}\right) \ln\left(\frac{0.500}{0.450}\right)$$

 $r_w = 0.00006 \text{ hr ft}^2 \text{ °F/Btu}$ 

The fins are parallel perforated aluminum plates with the tubes passing through the perforations.

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 $A_{FP} = (78.0)(15.75) - (460)\left(\frac{\pi}{4}\right)(0.500)^2$ 

$$A_{FP} = 1138.18 \ in^2 = 7.904 \ ft^2$$

The total fin area is

$$A_f = (2)(1060)(7.904)$$

$$A_f = 16,757 \ ft^2$$

The effective length of the tubes is 106.125 inches. The bare length of the tubes is

and a thickness of 0.010 inch. The available face area of each plate is

$$L_b = 106.125 - (1060)(0.010) = 95.525$$
 inches

The bare surface area of the tubes is

$$A_b = (460)(\pi)(0.500)(95.525)$$

$$A_b = 69,023 \ in^2 = 479 \ ft^2$$

Therefore,

$$A_o = 16,757 + 479 = 17,236 ft^2$$

The inside surface area of the tubes is

$$A_i = (460)(\pi)(0.450)(106.125)$$

$$A_i = 69,014 \ in^2 = 479 \ ft^2$$

From page 2-4 of Reference 3, the engine cooling water flow rate is 1100 gpm. From Reference 1, Oyster Creek does not use anti-freeze; therefore, this analysis will use the physical properties of fresh water. From Reference 2, the radiator was sized for a maximum engine water outlet

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	an ambient temperature of 90 F. From Table 13 (page A14)		an average	
$\rho_C = 61.2 \text{ lb/ft}^3$				
$(C_p)_C = 0.999 \text{ Btu/lb °F}$				
$k_{C} = 0.378$ Btu/hr ft °F				
$\mu_C = 2.98 \times 10^{-4} \text{ lb/ft s}$				
$Pr_{C} = 2.73$				
The velocity of the coolant	flowing through the tubes is			
$V_C = \left[\frac{1100 \text{ gal/min}}{(460)}\right] \left(\frac{1 \text{ m}}{60}\right)$	$\frac{\sin}{s} \left( \frac{1 \ ft^3}{7.4805 \ gal} \right) \left[ \frac{4}{\pi (0.450/12)} \right]$	$\frac{1}{(ft)^2} = 4.824 \ ft/s$		
The Reynolds number is				
$\operatorname{Re}_{c} = \frac{\left(61.2 \ lb \ ft^{3}\right)\left(4.824}{2.98 \times 10^{-5}}$	$\frac{ft/s}{(0.450/12 \ ft)} = 3.715X1$	0⁴		
The inside heat transfer coe	efficient is given by Equation (	5.63 (page 413) of Refer	ence 4	
$h_i = (0.023) \left(\frac{k_c}{ID}\right) (\operatorname{Re}_c)^{0.8} (\operatorname{Re}_c)^{0.8}$	$(9r_c)^{0.3} = (0.023) \left( \frac{0.378}{0.450/12} \right) (3.7)$	$15X10^4)^{0.8}(2.73)^{0.3}$		
$h_i = 1419 Btu/hr ft^2 °F$				
From Reference 1, the design Reference 4, at standard co	gn air flow rate is 125,000 scfi nditions (14.7 psia, 60 °F),	n. From Table 27 (page	A26) of	
$\rho_A=0.0739~lb/ft^3$				

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 $k_A = 0.0143$  Btu/hr ft °F

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$\mu_A = 1.214 X 10^{-5} \text{ lb/ft s}$	· · · · · · · · · · · · · · · · · · ·		
$Pr_A = 0.71$ The face area of the radia	tor is		
$A_F = (78.0)(106.125) = 82$	$277.75 in^2 = 57.484 ft^2$		
The face velocity of the a	ir is		
$V_{\infty} = \left[\frac{G_{A} ft^{3} / \min}{(57.484 ft^{2})}\right] \left(\frac{1 \text{ m}}{60}\right)$	$\left(\frac{\ln s}{s}\right)$		
$V_{\infty} = \frac{G_A}{3449.04} ft/s, \ G_A$	in scfm		
The tubes are arranged in estimated as follows:	a staggered triangular array, a	s shown in Figure C-1. S	$_{\rm T}$ and S <sub>L</sub> are
$S_T = \frac{39 \text{ inches}}{26} = 1.50 \text{ inc}$	ches		
$S_L = \frac{15.75 \text{ inches}}{9} = 1.75$	inches		
$a = \frac{1}{2}(S_T + OD) = \frac{1}{2}(1.50)$	+0.50) = 1.00 inch		
$b = \sqrt{\left(\frac{S_T}{2}\right)^2 + S_L^2} = \sqrt{\left(\frac{1.5}{2}\right)^2}$	$\left(\frac{10}{100}\right)^2 + (1.75)^2 = 1.90$ inches		
Since a is less than b, the 473 and 474 of Reference	maximum velocity of the air f 4)	lowing over the tubes is g	iven by (pages
$V_{\max} = \left(\frac{S_T}{S_T - OD}\right) V_{\infty} = \left(\frac{1}{1}\right)$	$\frac{1.50}{.50-0.50} \left( \frac{G_{A}}{3449.04} \right)$		
$V_{\rm max} = \frac{G_A}{2299.36} ft/s, G_A$	in sofm		

.....

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 $V_{\text{max}} = 54.36 \text{ ft/s}$ , at design flow of 125,000 scfm

The Reynolds number is given by

$$\operatorname{Re}_{A} = \frac{\rho_{A}V_{\max}(OD)}{\mu_{A}} = \frac{\left(0.0739\,lb\,/\,ft^{3}\right)\left(\frac{G_{A}}{2299.36}\,ft\,/\,s\right)\left(0.50/12\,ft\right)}{1.214X10^{-5}\,lb\,/\,ft\,s}$$

 $\operatorname{Re}_{A} = (0.11031)G_{A}$ ,  $G_{A}$  in scfm

 $\operatorname{Re}_{\lambda} = 1.379 X 10^4$ , at design flow of 125,000 scfm

Neglecting the correction factor for film temperature, the outside heat transfer coefficient is given by Equation 7.30 on page 475 of Reference 4,

$$h_{o} = (0.35) \left(\frac{k_{A}}{OD}\right) \left(\frac{S_{T}}{S_{L}}\right)^{0.2} (\text{Re}_{A})^{0.60} (\text{Pr}_{A})^{0.36} = (0.35) \left(\frac{0.0143}{0.50/12}\right) \left(\frac{1.5}{1.75}\right)^{0.2} (0.11031G_{A})^{0.60} (0.71)^{0.36}$$
  
$$h_{o} = (2.7431X10^{-2}) G_{A}^{0.60} Btu / hr ft^{2} \, {}^{\circ}F, \ G_{A} \text{ in scfm}$$
  
$$h_{o} = 31.36 Btu / hr ft^{2} \, {}^{\circ}F, \ \text{at design flow of } 125,000 \text{ scfm}$$

Fin efficiency will be calculated assuming equivalent circular fins, as follows,

$$R_i = \frac{OD}{2} = \frac{0.500}{2} = 0.250 \text{ in}$$

t = 0.010 inch

The fin face area associated with each tube is

$$A_{fF} = \frac{A_{FP}}{460}$$

This area is given by

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$\pi (R_o^2 - R_i^2) = \frac{A_{FP}}{460}$ , or			
$R_{o} = \left[\frac{A_{FP}}{460\pi} + R_{i}^{2}\right]^{0.5} = \left[\frac{1}{2}\right]^{0.5}$	$\frac{138.18}{460\pi} + (0.250)^2 \bigg]^{0.5} = 0.922 \ in$	n	
From Table 10 (page A9)	of Reference 4,		
k = 94.7 Btu/hr ft °F, alu	ninum		
$P = \left(R_o + \frac{t}{2} - R_i\right)^{3/2} \sqrt{\frac{t}{kt}}$	$\frac{2h_o}{R_o-R_i}$		
$P = \left(0.922 + \frac{0.010}{2} - 0.2\right)$	$50\right)^{3/2} \sqrt{\frac{2(31.36/12)}{(94.7)(0.010)(0.922 - 0.22)}}$	250)	
<i>P</i> = 1.60			
$Z = \frac{R_o + \left(\frac{t}{2}\right)}{R_i} = \frac{0.922 + \left(\frac{t}{2}\right)}{0.22}$	$\frac{0.010}{2}\right)$		
Z = 3.7			

From Figure 2.19 (page 107) of Reference 4, the fin efficiency is estimated to be

$$E_{f} = 0.40$$

This is the fin efficiency at design air flow (125,000 scfm). For this analysis it will be assumed that the fin efficiency remains constant at the design value during the heat up transient. Therefore,

$$U = \left[\frac{1}{(0.40)(2.7431X10^{-2}G_A^{0.60})} + \frac{0.001}{0.40} + 0.00006 + \left(\frac{17,236}{479}\right)\left(\frac{1}{1419} + 0.001\right)\right]^{-1}$$

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$$U = \frac{15.649}{\left[1 + \frac{1426.22}{G_A^{560}}\right]} Btu / hr ft^2 * F$$
, G<sub>A</sub> in soft $U = 6.963 Btu / hr ft^2 * F$ , at design flow of 125,000 softTherefore, $(UA)_R = (17,236) \frac{15.649}{\left[1 + \frac{1426.22}{G_A^{560}}\right]}$ , $(UA)_R = \frac{2.697 \times 10^5}{\left[1 + \frac{1426.22}{G_A^{560}}\right]}$  $(UA)_R = \frac{2.697 \times 10^5}{\left[1 + \frac{1426.22}{G_A^{560}}\right]}$  $(UA)_R = \frac{2.000 \times 10^5}{\left[1 + \frac{1426.22}{G_A^{560}}\right]}$  $(UA)_R = 1.200 \times 10^5 Btu / hr * F$ , at design flow of 125,000 softFrom Equation 18, the final steady state temperature of the coolant exiting the engine (Tc2) is $T_{c2f} = T_{al} + \left[\frac{1}{2C_A} + \frac{1}{(UA)_R} + \frac{1}{2C_C}\right] Q_T$ ; or

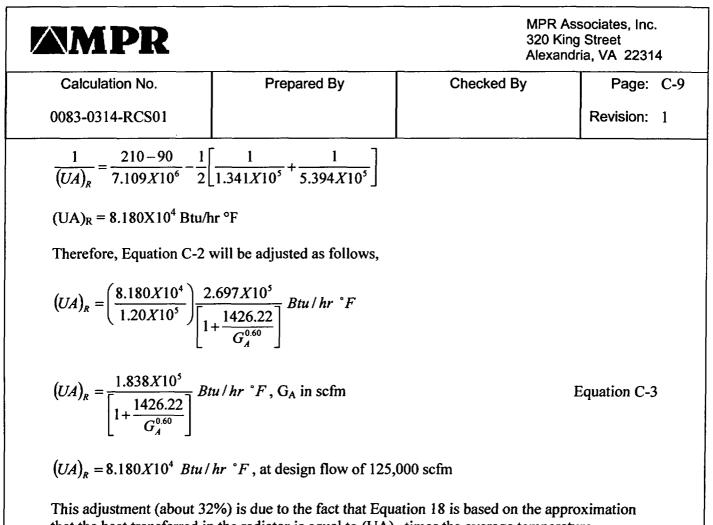
 $\frac{1}{(UA)_R} = \frac{T_{C2f} - T_{Ai}}{Q_T} - \frac{1}{2} \left[ \frac{1}{C_A} + \frac{1}{C_C} \right]$ 

From Reference 2, the radiator was sized for a maximum engine water outlet temperature of 210 °F with an ambient temperature of 90 °F. From Appendices A and B,

$$Q_T = 7.109X10^6 Btu / hr$$

$$C_C = 5.394X10^5 Btu / hr °F$$

$$C_A = 1.341X10^5 Btu / hr °F, \text{ at design flow}$$



that the heat transferred in the radiator is equal to  $(UA)_R$  times the average temperature difference between the coolant and the air. In reality, the heat transferred is equal to  $(UA)_R$  times the corrected log mean temperature difference.

From Equation 23, the final steady state temperature of the coolant entering the engine (T<sub>C1</sub>) is

$$T_{C1f} = T_{Ai} + \left[\frac{1}{2C_{A}} + \frac{1}{(UA)_{R}} - \frac{(Q_{E} - Q_{L})}{2C_{C}Q_{T}}\right]Q_{I}$$

From Appendix A,

$$Q_{E} = 5.123X10^{6} Btu / hr$$

$$Q_{L} = 1.986X10^{6} Btu / hr$$

$$Q_{T} = 7.109X10^{6} Btu / hr$$

$$T_{C1f} = 90.0 + \left[\frac{1}{2(1.341X10^{5})} + \frac{1}{8.180X10^{4}} - \frac{(5.123X10^{6} - 1.986X10^{6})}{2(5.394X10^{5})(7.109X10^{6})}\right] (7.109X10^{6})$$

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$T_{C1f} = 200.51 \ ^{\circ}F$				
<i>(UA)c</i> The lube oil cooler design	information used in the follo	wing analysis is from Refe	erence 7.	
The lube oil cooler is a fir the insides of the tubes an	nned tube cross flow heat excl d the lube oil flowing over th ace 6, the overall heat transfer	nanger with the cooling wa	ater flowing on of the tubes.	
$U = \left[\frac{1}{E_f h_o} + \frac{r_o}{E_f} + r_w + \left(\frac{A_o}{A_o}\right)\right]$	$\frac{A_o}{A_i}\left(\frac{1}{h_i}+r_i\right)\right]^{-1}$		Equation C-1	
Where,				
U = the overall heat transf	er coefficient, Btu/hr ft <sup>2</sup> °F			
$h_o = film$ heat transfer coe	fficient on outside surfaces of	tubes, Btu/hr ft <sup>2</sup> °F		
	fficient on inside surfaces of t	ubes, Btu/hr ft <sup>2</sup> °F		
$h_i = film$ heat transfer coefficients	inclose on malde surfaces of t	,		
	ance on outside surfaces of tu			
		bes, hr ft <sup>2</sup> °F/Btu		
$r_o =$ thermal fouling resist	ance on outside surfaces of tu	bes, hr ft <sup>2</sup> °F/Btu		
$r_o =$ thermal fouling resistar $r_i =$ thermal fouling resista	ance on outside surfaces of tu	bes, hr ft <sup>2</sup> °F/Btu		
$r_o =$ thermal fouling resista $r_i =$ thermal fouling resista $r_w =$ thermal resistance of	ance on outside surfaces of tu unce on inside surfaces of tube tube walls, hr ft <sup>2</sup> °F/Btu	bes, hr ft <sup>2</sup> °F/Btu		
$r_o =$ thermal fouling resista $r_i =$ thermal fouling resistance $r_w =$ thermal resistance of $E_f =$ fin efficiency	ance on outside surfaces of tu unce on inside surfaces of tube tube walls, hr ft <sup>2</sup> °F/Btu nsfer surface area, ft <sup>2</sup>	bes, hr ft <sup>2</sup> °F/Btu		

 $r_o = 0.001$  hr ft<sup>2</sup> °F/Btu, engine lube oil

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 $r_i = 0.001$  hr ft<sup>2</sup> °F/Btu, engine jacket water

The cooler contains 1384 red brass tubes with an OD of 0.250 inch and a wall thickness of 0.017 inch. Therefore, the ID of each tube is 0.216 inch (0.250-2x0.017). From page 125 of Reference 6, the thermal resistance of the tube walls is,

$$r_{w} = \left(\frac{OD}{24k_{t}}\right) \ln\left(\frac{OD}{ID}\right)$$

Where,

 $k_t$  = thermal conductivity of the tube material, Btu/hr ft °F

From Table 10 (page A9) of Reference 4,

 $k_t = 35.2$  Btu/hr ft °F, red brass

Therefore,

$$r_{w} = \left(\frac{0.250}{24(35.2)}\right) \ln\left(\frac{0.250}{0.216}\right)$$

 $r_w = 0.00004 \text{ hr ft}^2 \text{ °F/Btu}$ 

The fins are parallel perforated aluminum plates with the tubes passing through the perforations. The cooler contains 784 plates, each with a height of 32.0 inches, a width of 6.0 inches and a thickness of 0.010 inch. The available face area of each plate is

$$A_{FP} = (32.0)(6.0) - (1384) \left(\frac{\pi}{4}\right) (0.250)^2$$
$$A_{FP} = 124.06 \text{ in}^2 = 0.8616 \text{ ft}^2$$

The total fin area is

$$A_f = (2)(784)(0.8616)$$

$$A_f = 1351 ft^2$$

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The effective length of the	tubes is 26.66 inches. The t	are length of the tubes is	
$L_b = 26.66 - (784)(0.010)$	=18.82 inches		
The bare surface area of the	e tubes is		
$A_b = (1384)(\pi)(0.250)(18.8)$	22)		
$A_b = 20,457 \ in^2 = 142 \ ft^2$			
Therefore,			
$A_o = 1351 + 142 = 1493 ft^2$			
The inside surface area of	the tubes is		
$A_i = (1384)(\pi)(0.216)(26.6)$	6)		
$A_i = 25,038 in^2 = 174 ft^2$			

The velocity of the coolant flowing through the tubes is

$$V_{C} = \left[\frac{1100 \ gal / \min}{(1384)}\right] \left(\frac{1 \ \min}{60 \ s}\right) \left(\frac{1 \ ft^{3}}{7.4805 \ gal}\right) \left[\frac{4}{\pi (0.216/12 \ ft)^{2}}\right] = 6.959 \ ft/s$$

The Reynolds number is

$$\operatorname{Re}_{c} = \frac{\left(61.2 \ lb \ ft^{3}\right)\left(6.959 \ ft \ s\right)\left(0.216 \ l2 \ ft\right)}{2.98 \times 10^{-4} \ lb \ ft \ s} = 2.5725 \times 10^{4}$$

The inside heat transfer coefficient is given by Equation 6.63 (page 413) of Reference 4

$$h_{i} = (0.023) \left(\frac{k_{c}}{ID}\right) (\text{Re}_{c})^{0.8} (\text{Pr}_{c})^{0.4} = (0.023) \left(\frac{0.378}{0.216/12}\right) (2.5725X10^{4})^{0.8} (2.73)^{0.4}$$

 $h_i = 2436.1 Btu / hr ft^2$ °F

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pump flow rate). From R oil outlet temperature of 2	of Reference 3, the engine lube deference 2, the lube oil cooler 240 °F with an ambient tempe bout 165 °F. From Table 16 (j	s were sized for a maximu rature of 90 °F. Therefore	m engine lube	
$(C_p)_L = 0.503 \text{ Btu/lb °F}$				
$k_L = 0.0801$ Btu/hr ft °F				
$\mu_L = 2.982 \times 10^{-2} \text{ lb/ft s}$				
$Pr_{L} = 661$				
The face area of the coole	er is			
$A_F = (32.0)(26.66) = 853.$	$12 in^2 = 5.924 ft^2$			
The face velocity of the lu	ube oil is			
$V_{\infty} = \left[\frac{390 \text{ gal}/\min}{(5.924 \text{ ft}^2)}\right] \left(\frac{1 \text{ m}}{60}\right)$	$\frac{\min}{0.s} \left( \frac{1 ft^3}{7.4805 gal} \right)$			
$V_{\infty} = 0.1467 \ ft/s$				
The tubes are arranged in estimated as follows:	a staggered triangular array, a	us shown in Figure C-1. S <sub>1</sub>	$_{\rm f}$ and ${\rm S}_{\rm L}$ are	
$S_T = \frac{32 \text{ inches}}{87} = 0.3678 \text{ inches}$	inch			
$S_L = \frac{6}{16} = 0.3750$ inch				

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$$b = \sqrt{\left(\frac{S_T}{2}\right)^2 + S_L^2} = \sqrt{\left(\frac{0.3678}{2}\right)^2 + (0.3750)^2} = 0.4177 \text{ inches}$$

Since a is less than b, the maximum velocity of the lube oil flowing over the tubes is given by (pages 473 and 474 of Reference 4)

$$V_{\max} = \left(\frac{S_T}{S_T - OD}\right) V_{\infty} = \left(\frac{0.3678}{0.3678 - 0.250}\right) (0.1467)$$

$$V_{\rm max} = 0.4580 \ ft/s$$

The Reynolds number is given by

$$\operatorname{Re}_{L} = \frac{\rho_{L} V_{\max}(OD)}{\mu_{L}} = \frac{(53.4 \, lb \, / \, ft^{3})(0.4580 \, ft \, / \, s)(0.250 / 12 \, ft)}{2.982 X 10^{-2} \, lb \, / \, ft \, s}$$

 $Re_L = 17.087$ 

Neglecting the correction factor for film temperature, the outside heat transfer coefficient is given by Equation 7.28 on page 475 of Reference 4,

$$h_o = (0.9) \left(\frac{k_L}{OD}\right) (\text{Re}_L)^{0.4} (\text{Pr}_L)^{0.36} = (0.9) \left(\frac{0.0801}{0.250/12}\right) (17.087)^{0.40} (661)^{0.36}$$
$$h_o = 111.55 \ Btu \ / \ hr \ ft^2 \ °F$$

Fin efficiency will be calculated assuming equivalent circular fins, as follows,

$$R_i = \frac{OD}{2} = \frac{0.2500}{2} = 0.1250 \text{ in}$$

t = 0.010 inch

The fin face area associated with each tube is

$$A_{fF} = \frac{A_{FP}}{1384}$$

This area is given by

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$\pi (R_o^2 - R_i^2) = \frac{A_{FP}}{1384}$ , or			
$R_{o} = \left[\frac{A_{FP}}{1384\pi} + R_{i}^{2}\right]^{0.5} = \left[$	$\frac{124.06}{1384\pi} + (0.1250)^2 \bigg]^{0.5} = 0.210$	in	
From Table 10 (page A9)	of Reference 4,		
k = 94.7 Btu/hr ft °F, alur	ninum		
$P = \left(R_o + \frac{t}{2} - R_i\right)^{3/2} \sqrt{\frac{t}{kt(t)}}$	$\frac{2h_o}{R_o - R_i}$		
$P = \left(0.210 + \frac{0.010}{2} - 0.12\right)$	$(250)^{3/2} \sqrt{\frac{2(111.55/12)}{(94.7)(0.010)(0.210-0)}}$	.1250)	
<i>P</i> = 0.41			
$Z = \frac{R_o + \left(\frac{t}{2}\right)}{R_i} = \frac{0.210 + \left(\frac{t}{2}\right)}{0.12}$	$\frac{0.010}{2}$		
<i>Z</i> = 1.72			
From Figure 2.19 (page 1	07) of Reference 4, the fin effic	eiency is estimated to be	
$E_{f} = 0.93$			
$U = \left[\frac{1}{(0.93)(111.55)} + \frac{0.0}{0.9}\right]$	$\frac{01}{3} + 0.00004 + \left(\frac{1493}{174}\right)\left(\frac{1}{2436.1}\right)$	$+0.001\Big]^{-1}$	
$U = 43.824 Btu / hr ft^2 °F$	,		
$(UA)_{C} = (43.824)(1493)$			
$(UA)_{c} = 6.543 X 10^{4} Btu/$	hr °F		Equation C-4

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From Equations 41 and 42, the final steady state temperature of the lube oil exiting the engine  $(T_{L2})$  is

$$T_{L2f} = T_{C1f} + \left[\frac{1}{2C_L} + \frac{1}{(UA)_C} - \frac{1}{2C_C}\right]Q_L$$
, or

 $\frac{1}{(UA)_{C}} = \frac{T_{L2f} - T_{C1f}}{Q_{L}} + \frac{1}{2} \left[ \frac{1}{C_{C}} - \frac{1}{C_{L}} \right]$ 

From Reference 2, the cooler was sized for a maximum engine lube oil outlet temperature of 240 °F with an ambient temperature of 90 °F. From Appendices A and B,

$$Q_L = 1.986X10^6 Btu / hr$$

 $C_c = 5.394 X 10^5 Btu / hr °F$ 

 $C_L = 8.402 X 10^4 Btu / hr °F$ 

$$\frac{1}{(UA)_C} = \frac{240.0 - 200.51}{1.986 \times 10^6} + \frac{1}{2} \left[ \frac{1}{5.394 \times 10^5} - \frac{1}{8.402 \times 10^4} \right]$$

 $(UA)_{C} = 6.729 \times 10^{4} \text{ Btu/hr }^{\circ}\text{F}$ 

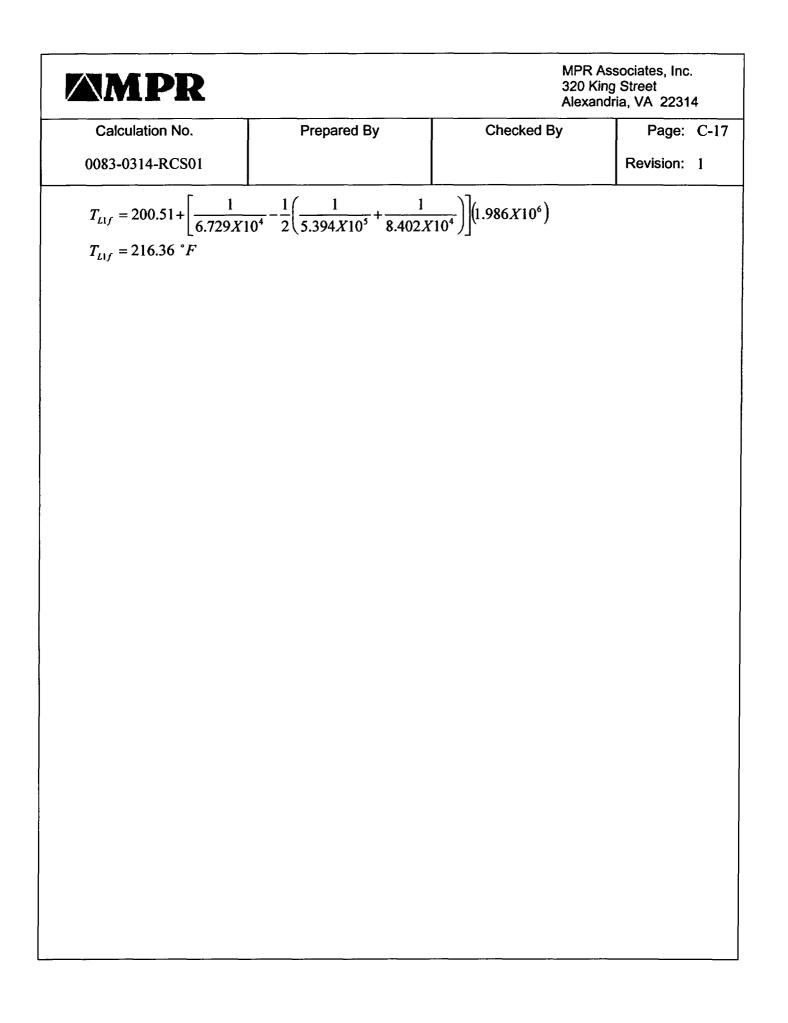
Therefore,  $(UA)_C$  will be adjusted to,

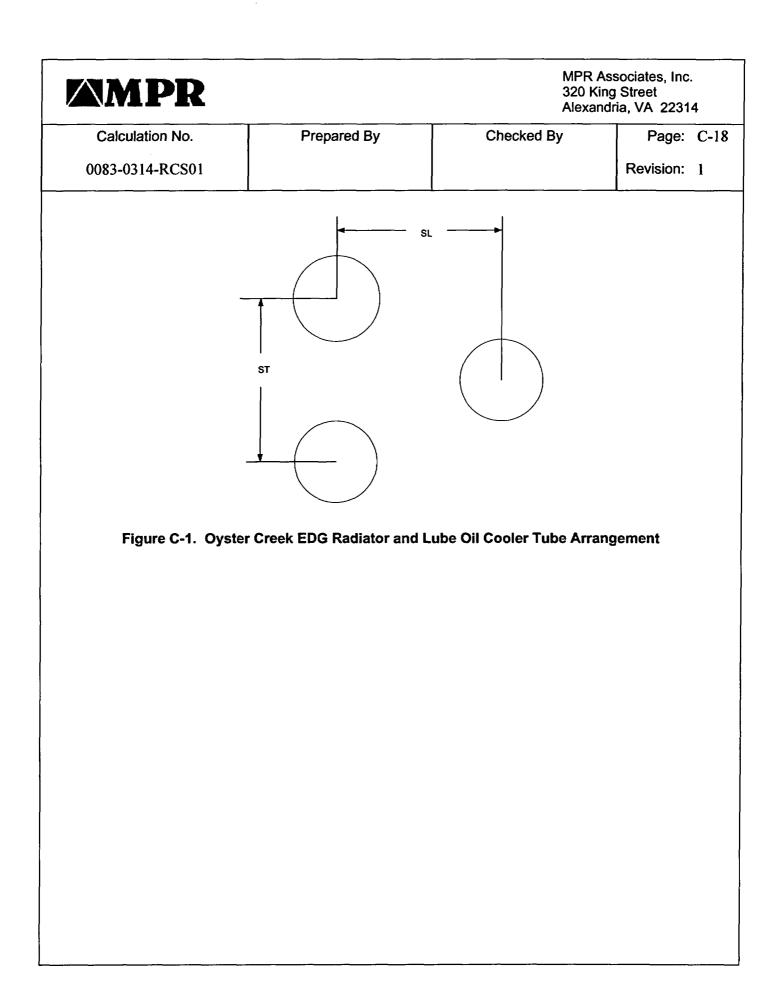
Equation C-5

This adjustment (about 3%) is due to the fact that Equations 41 and 42 are based on the approximation that the heat transferred in the cooler is equal to  $(UA)_C$  times the average temperature difference between the coolant and the lube oil. In reality, the heat transferred is equal to  $(UA)_C$  times the corrected log mean temperature difference.

From Equation 43, the final steady state temperature of the lube oil entering the engine (T<sub>L1</sub>) is

$$T_{L1f} = T_{C1f} + \left[\frac{1}{(UA)_{C}} - \frac{1}{2}\left(\frac{1}{C_{C}} + \frac{1}{C_{L}}\right)\right]Q_{L}$$





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[(mass)X(specific heat)] f	dix is to determine the heat all or the cooling water system u sed in this appendix are listed	sed in the Oyster Creek EI	DG cooling	
The purpose of this appen [(mass)X(specific heat)] for system. The references us	or the cooling water system u	sed in the Oyster Creek EI	DG cooling	

The heat up data for the coolant and lube oil are shown in Figure D-1. From Equation 26 the coolant temperatures shown in Figure D-1 can be represented by the following equation,

$$T_C = T_{Cf} - \left(T_{Cf} - T_{Ci}\right)e^{-\lambda_{CI}t}$$

Where,  $\lambda_{CT}$  is the time decay constant for the coolant system.

Assuming the time decay constant ( $\lambda_{LT}$ ) for the lube oil system is approximately equal to  $\lambda_{CT}$ , the lube oil temperatures shown in Figure D-1 can be represented by the following equation,

$$T_L = T_{Lf} - (T_{Lf} - T_{Li})e^{-\lambda_L t^t}$$

In Equations D-1 and D-2, the subscript f refers to final conditions, the subscript i refers to initial conditions and the subscript T refers to test conditions.

Equations D-1 and D-2 can also be written as,

**Equation D-2** 

**Equation D-1** 

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$\ln\left(\frac{T_{Cf} - T_{Ci}}{T_{Cf} - T_{C}}\right) = \lambda_{CT}t$		Eq	uation D-3
$\ln\!\left(\frac{T_{Lf}-T_{Li}}{T_{Lf}-T_{L}}\right) = \lambda_{LT}t$		Eq	uation D-4
From the test data,			
$T_{Ci} = 141 \text{ °F}$			
T <sub>Li</sub> = 135 °F			
From the test data, it appe for the coolant and lube o	ars that the final steady state to il, respectively. That is,	emperatures are about 18	7 °F and 231 °F
T <sub>Cf</sub> = 187 °F			
$T_{Lf} = 231  {}^{\circ}F$			
Therefore, Equations D-3	and D-4 become,		
$\ln\!\left(\frac{46}{187 - T_C}\right) = \lambda_{CT} t$		Equ	nation D-5
$\ln\!\left(\frac{96}{231\!-\!T_L}\right) = \lambda_{LT} t$		Εqι	ation D-6
The measured values of T results plotted as a function	$_{\rm C}$ and ${\rm T_L}$ have been substituted on of time, see Figure D-2.	l into Equations D-5 and 1	D-6, and the
data fall on essentially the greater than 30 minutes, the thermostatic temperature of	e seen that for the first 30 minutes same line, indicating that $\lambda_{LT}$ here is considerable scatter in the control value regulating the flow at their final steady state values.	is approximately equal to he data. This is believed w rate of coolant through	$\lambda_{CT}$ . For times to be due to the
The coolant and lube oil d	ata shown in Figure D-2 for th	e first 30 minutes are con	nhined to

The coolant and lube oil data shown in Figure D-2 for the first 30 minutes are combined to determine an effective value for  $\lambda_{CT}$ . A linear regression analysis of these data gives,  $\lambda_{CT} = 8.836 \times 10^{-2}$ /min

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From Equation 13,

$$\lambda_{CT} = \frac{(UA)_{RT}}{\left(MC_p\right)_{CTT} \left[1 + (UA)_{RT}/2C_{AT}\right]}, \text{ or}$$

$$\left(MC_p\right)_{CTT} = \frac{(UA)_{RT}}{\lambda_{CT} \left[1 + (UA)_{RT}/2C_{AT}\right]}$$
Equation D-7

Reference 9 provides design information for the MP-36 radiator; however, sufficient information is not provided to calculate  $(UA)_{RT}$  from first principles; therefore,  $(UA)_{RT}$  and  $C_{AT}$  will be estimated from  $(UA)_R$  and  $C_A$  given in Appendices B and C (at design air flow rate of 125,000 scfm) as follows.

$$(UA)_{RT} = \left(\frac{16}{20}\right) (UA)_{R} = \left(\frac{16}{20}\right) (8.180X10^{4}) = 6.544X10^{4} Btu / hr °F$$
$$C_{AT} = \left(\frac{16}{20}\right) C_{A} = \left(\frac{16}{20}\right) (1.341X10^{5}) = 1.073X10^{5} Btu / hr °F$$

Therefore,

$$\left(MC_{p}\right)_{CTT} = \frac{6.544X10^{4}}{\left(8.836X10^{-2}\left[1 + \frac{6.544X10^{4}}{2\left(1.073X10^{5}\right)}\right]} \left(\frac{1hr}{60\min}\right) = 9.459X10^{3} Btu/^{\circ}F$$

Assuming (MC<sub>p</sub>) is proportional to engine size,

$$(MC_p)_{CT} = \left(\frac{20}{16}\right)(MC_p)_{CTT} = \left(\frac{20}{16}\right)(9.459X10^3) = 1.182X10^4 Btu/^{\circ}F$$

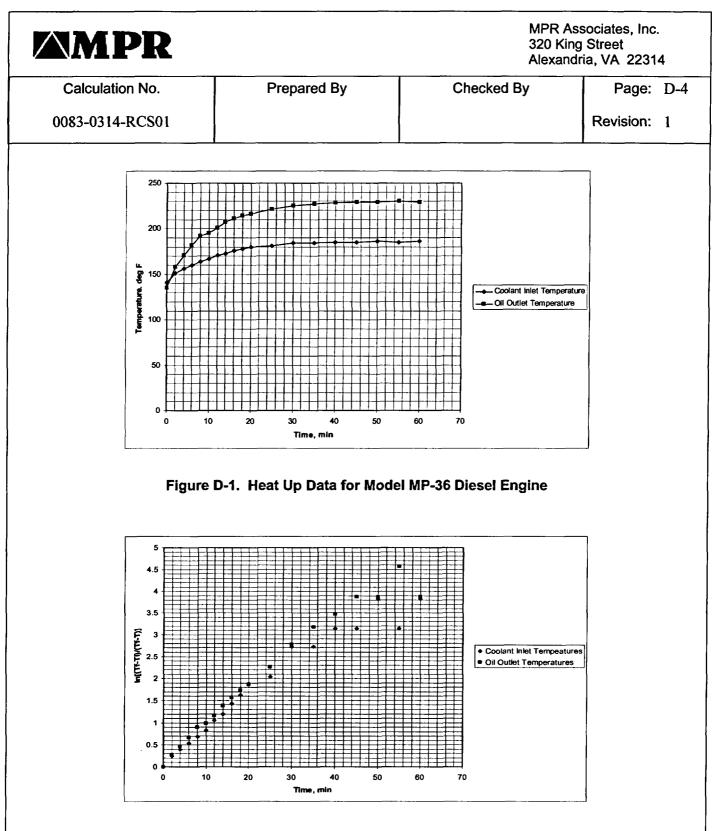


Figure D-2. Heat Up Data for Model MP-36 Diesel Engine