

Appendix 5 Structural Integrity of Spent Fuel Pool Structures
Subject to Aircraft Crashes

The generic data provided in [Ref. 1] was used to assess the likelihood of an aircraft crash into or near a decommissioned spent fuel pool. Aircraft damage can affect the structural integrity of the spent fuel pool or affect the availability of nearby support systems, such as power supplies, heat exchangers and water makeup sources, and may also affect recovery actions.

The frequency of an aircraft crashing into a site, F , was obtained from the four-factor formula in DOE-STD-3014-96, and is referred to as the effective aircraft target area model:

$$F = \sum_{i,j,k} N_{ijk} \cdot P_{ijk} \cdot f_{ijk}(x, y) \cdot A_{ij} \quad \text{Equation A5-1}$$

where:

- N_{ijk} = estimated annual number of site-specific aircraft operations (no./yr)
- P_{ijk} = aircraft crash rate per takeoff and landing for near-airport phases and per flight for in-flight (nonairport) phase of operation
- $f_{ijk}(x, y)$ = aircraft crash location probability (per square mile)
- A_{ij} = site-specific effective area for the facility of interest, including skid and fly-in effective areas (square miles)
- i = (index for flight phase): $i=1,2$, and 3 (takeoff, in-flight, landing)
- j = (index for aircraft category, or subcategory)
- k = (index for flight source): there could be multiple runways and nonairport operations

The site-specific area is shown in Figure A5-1 and is further defined as:

and where:

$$A_{\text{eff}} = A_f + A_s \quad \text{where } A_{\text{eff}} = \text{total effective target area} \quad H = \text{height of facility}$$

where:

$$A_f = (WS + R) \cdot (H \cdot \cot\theta) + \frac{2 \cdot L \cdot W \cdot WS}{R} + L \cdot W$$

$$A_s = (WS + R) \cdot S$$

Equation A5-2

- A_f = effective fly-in area
- A_s = effective skid area
- WS = wing span
- $\cot\theta$ = mean of cotangent of aircraft impact angle
- L = length of facility
- W = width of facility
- S = aircraft skid distance
- R = length of facility diagonal

Alternatively, a point target area model was defined as the area (length times width) of the facility in question.

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Table A5-1 summarizes the generic aircraft data and crash frequency values for five aircraft types (from Tables B-14 through B-18 of DOE-STD-3014-96).

The data given in Table A5-1 was used to determine the frequency of aircraft hits per year for various building sizes (length, width, and height) for the minimum, average, and maximum crash rates. The resulting frequencies are given in Table A5-2. The product $N_{ijk} * P_{ijk} * f_{ijk}(x,y)$ for Equation A5-1 was taken from the crashes per mi^2 -yr and A_{ij} was obtained from Equation A5-2 based on the aircraft characteristics. Two sets of data were generated: one considered the wing and skid lengths using the effective aircraft target area model and a second case which considered only the area (length times width) of the site using the point target area model.

The building characteristics were chosen to cover a range of a typical spent fuel pool to that of the PWR auxiliary buildings or the BWR secondary containment structure. The results from the DOE effective aircraft target area model, using the generic data in Table A5-1, were compared to the results of two evaluations reported in [Ref. 2]

The first evaluation of aircraft crash hits was summarized by C.T. Kimura et al [Ref. 3]. DWTF Building 696 was assessed. It was a 254-foot-long by 80-foot wide, 1-story, 39-foot high structure. The results of Kimura's study are shown in Table A5-3.

Applying the DOE generic data to the DWTF resulted in a frequency range of 6.6×10^{-5} hits per year to 6.5×10^{-9} hits per year, with an average value of 4.4×10^{-6} per year, for the effective aircraft target area model. For the point target area model, the range was 4.4×10^{-10} to 2.2×10^{-6} per year, with an average value of 1.5×10^{-7} per year.

The second evaluation was presented in a paper by K. Jamali, et al., "Application of Aircraft Crash Hazard Assessment Methods to Various Facilities in the Nuclear Industry," in which additional facility evaluations were summarized. For the Seabrook Nuclear Power Station, Jamali's application of the DOE effective aircraft target area model to the final safety analysis report (FSAR) data resulted in an impact frequency 2.38×10^{-5} per year. The Millstone 3 plant area was reported as 9.5×10^{-3} square miles and the FSAR aircraft crash frequency as 1.6×10^{-6} per year. Jamali applied the DOE effective aircraft target area model to information found in the Millstone Unit 3 FSAR. Jamali reported an impact frequency of 2.74×10^{-6} per year using the areas published in the FSAR and 2.31×10^{-5} per year and using the effective area calculated the effective aircraft target area model.

When the generic DOE data in Table A5-1 was used (for a 514x514x100-foot site), the estimated impact frequency range was 2.9×10^{-5} to 6.3×10^{-9} per year, with an average of 1.9×10^{-6} per year, for the point target area model. The effective aircraft target area model gave estimated range of 2.4×10^{-4} to 3.1×10^{-8} per year, with an average of 1.6×10^{-5} per year.

A site-specific evaluation for Three Mile Island Units 1 and 2 was documented in NUREG/CR-5042, "Evaluation of External Hazards to Nuclear Power Plants in the United States," Lawrence Livermore National Laboratory, December 1987. The NUREG estimated the aircraft crash frequency to be 2.3×10^{-4} accidents per year, about the same value as would be predicted with the DOE data set for the maximum crash rate for a site area of 0.01 square miles.

NUREG/CR-5042 summarized a study of a power plant response to aviation accidents. The

results are given in Table A5-4. The probability of the penetration of an aircraft through reinforced concrete was taken from that study.

There is reasonable assurance that the DOE model and generic data provide a range of aircraft crash hit frequencies that would be consistent with plant-specific evaluations. At this level of effort, the resulting damage from an aircraft crash cannot be fully evaluated on a plant-specific bases.

A detailed structural evaluation is beyond the scope of this effort. In general, PWR spent fuel pools are located on, or below grade, and BWR spent fuel pools, though generally elevated about 100 feet above grade, are inside a secondary containment structure. The vulnerability of support systems (power supplies, heat exchangers, and makeup water supplies) requires a knowledge of the size and location of these systems, information not readily available.

Calculated Values for Risk-informed Assessment of Spent Fuel Pool

Significant Pool Damage

PWRs The value for significant PWR spent fuel pool damage from a direct hit was estimated based on the point target model for a 100x50 foot pool with a conditional probability of 0.32 (large aircraft penetrating 6-ft of reinforced concrete) that the crash resulted in significant damage. If 1-of-2 aircraft are large and 1-of-2 crashes result in spent fuel uncover, then the estimated range is 4.3×10^{-8} to 9.6×10^{-12} per year. The average value was estimated to be 2.9×10^{-9} per year.

BWRs The value for significant BWR spent fuel pool damage resulting from a direct hit was estimated to be the same as for the PWR, 4.3×10^{-8} to 9.6×10^{-12} per year. The average value was estimated to be 2.9×10^{-9} per year. Mark-I and Mark-II secondary containments do not appear to have any significant structures to reduce the likelihood of penetration, although on one side there may be a reduced likelihood due to other structures. Mark-III secondary containments may reduce the likelihood of penetration, the spent fuel pool being protected by additional structures.

Support System Availability

The value for loss of a support system (power supply, heat exchanger or makeup water supply) was estimated based on the DOE model including wing and skid area for a 400x200x30-foot area with a conditional probability of 0.01 that one of these systems would be hit. The estimated value range was 1.0×10^{-6} to 1.0×10^{-10} per year. The average value was estimated to be 7.0×10^{-8} per year.

Alternatively, the value for loss of a support system was estimated based on the DOE model including wing and skid area for a 10x10x10-foot structure. The estimated value range was 1.1×10^{-5} to 1.1×10^{-9} per year with the wing and skid area modeled, with the average estimated to be 7.3×10^{-7} per year. Using the point model, the estimated value range was 1.1×10^{-8} to 2.4×10^{-12} per year without the wing and skid area modeled, with the average estimated to be 7.4×10^{-10} per year.

References:

- (1) DOE-STD-3014-96, "Accident Analysis for Aircraft Crash Into Hazardous Facilities," U.S. Department of Energy (DOE), October 1996.
- (2) A. Mosleh and R.A. Bari (Eds), "Probabilistic Safety Assessment and Management," PSAM 4, Volume 3, Proceedings of the 4th International Conference on Probabilistic Safety Assessment and Management, September 13-18, 1998, New York City.
- (3) C.T. Kimura, et al., "Aircraft Crash Hit Analysis of the Decontamination and Waste Treatment Facility", Lawrence Livermore National Laboratory (LLNL).

Table A5-1 Generic Aircraft Data

Aircraft	Wingspan (ft)	Skid distance (ft)	cotθ	Crashes per mi ² -yr			Notes:
				Min	Ave	Max	
General aviation	50	1440	10.2	1x10 ⁻⁷	2x10 ⁻⁴	3x10 ⁻³	
Air carrier	98	60	8.2	7x10 ⁻⁸	4x10 ⁻⁷	2x10 ⁻⁶	
Air taxi	58	60	8.2	4x10 ⁻⁷	1x10 ⁻⁶	8x10 ⁻⁶	
Large military	223	780	7.4	6x10 ⁻⁸	2x10 ⁻⁷	7x10 ⁻⁷	takeoff
Small military	100	447	10.4	4x10 ⁻⁸	4x10 ⁻⁶	6x10 ⁻⁸	landing

Table A5-2 Aircraft Hits Per Year

Building (L x W x H) (ft)	Average effective area (mi ²)	Minimum hits (per year)	Average hits (per year)	Maximum hits (per year)
With the DOE effective aircraft target area model				
100 x 50 x 30	6.9x10 ⁻³	3.2x10 ⁻⁹	2.1x10 ⁻⁶	3.1x10 ⁻⁵
200 x 100 x 30	1.1x10 ⁻²	5.3x10 ⁻⁹	3.7x10 ⁻⁶	5.5x10 ⁻⁵
400 x 200 x 30	2.1x10 ⁻²	1.0x10 ⁻⁸	7.0x10 ⁻⁶	1.0x10 ⁻⁴
200 x 100 x 100	1.8x10 ⁻²	9.6x10 ⁻⁹	5.1x10 ⁻⁶	7.6x10 ⁻⁵
400 x 200 x 100	3.3x10 ⁻²	1.8x10 ⁻⁸	9.6x10 ⁻⁶	1.4x10 ⁻⁴
80 x 40 x 30	6.1x10 ⁻³	2.8x10 ⁻⁹	1.8x10 ⁻⁶	2.7x10 ⁻⁵
10 x 10 x 10	2.9x10 ⁻³	1.1x10 ⁻⁹	7.3x10 ⁻⁷	1.1x10 ⁻⁵
With the point target area model				
100 x 50 x 0	1.8x10 ⁻⁴	1.2x10 ⁻¹⁰	3.7x10 ⁻⁸	5.4x10 ⁻⁷
200 x 100 x 0	7.2x10 ⁻⁴	4.8x10 ⁻¹⁰	1.5x10 ⁻⁷	2.2x10 ⁻⁶
400 x 200 x 0	2.9x10 ⁻³	1.9x10 ⁻⁹	5.9x10 ⁻⁷	8.6x10 ⁻⁶
80 x 40 x 0	1.1x10 ⁻⁴	1.1x10 ⁻¹¹	2.4x10 ⁻⁸	3.5x10 ⁻⁷
10 x 10	3.6x10 ⁻⁶	2.4x10 ⁻¹²	7.4x10 ⁻¹⁰	1.1x10 ⁻⁸

Table A5-3 DWTF Aircraft Crash Hit Frequency (Per Year)

Period	Air Carriers	Air Taxes	General Aviation	Military Aviation	Total ⁽¹⁾
1995	1.72×10^{-7}	2.47×10^{-6}	2.45×10^{-5}	5.03×10^{-7}	2.76×10^{-5}
1993-1995	1.60×10^{-7}	2.64×10^{-6}	2.82×10^{-5}	6.47×10^{-7}	3.16×10^{-5}
1991-1995	1.57×10^{-7}	2.58×10^{-6}	2.89×10^{-5}	7.23×10^{-7}	3.23×10^{-5}
1986-1995	1.52×10^{-7}	2.41×10^{-6}	2.89×10^{-5}	8.96×10^{-7}	3.23×10^{-5}

Note (1): Various periods were studied to assess variations in air field operations.

Table A5-4 Probability of Penetration as a Function of Location and Concrete Thickness

		Probability of penetration			
		Thickness of reinforced concrete			
Plant location	Aircraft type	1 foot	1.5 feet	2 feet	6 feet
≤ 5 miles from airport	Small ≤ 12,000 lbs	0.003	0	0	0
	Large > 12,000 lbs	0.96	0.52	0.28	0
> 5 miles from airport	Small ≤ 12,000 lbs	0.28	0.06	0.01	0
	Large > 12,000 lbs	1.0	1.0	0.83	0.32

Figure A5-1 - Rectangular Facility Effective Target Area Elements

